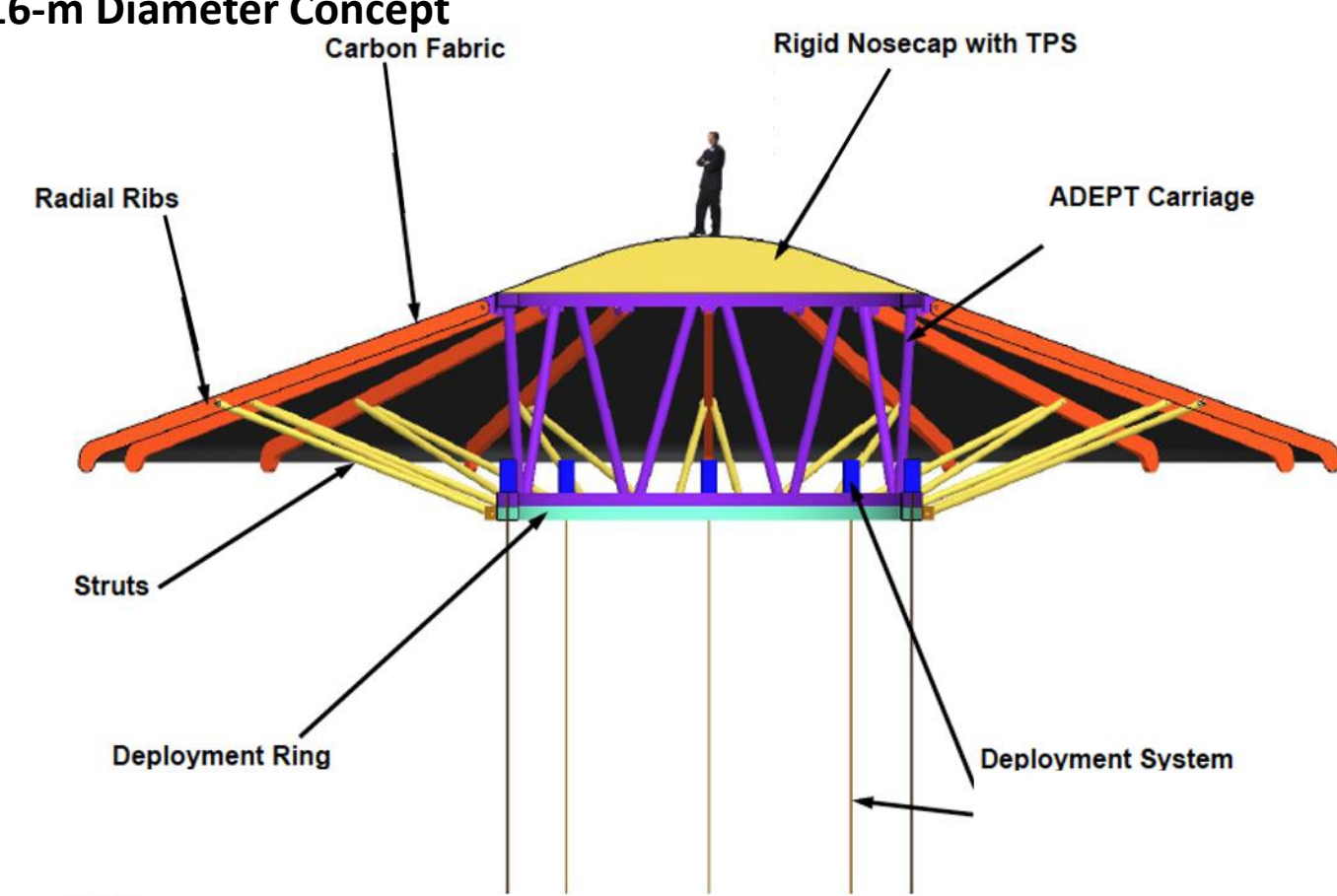


ADEPT 16-m Diameter Concept



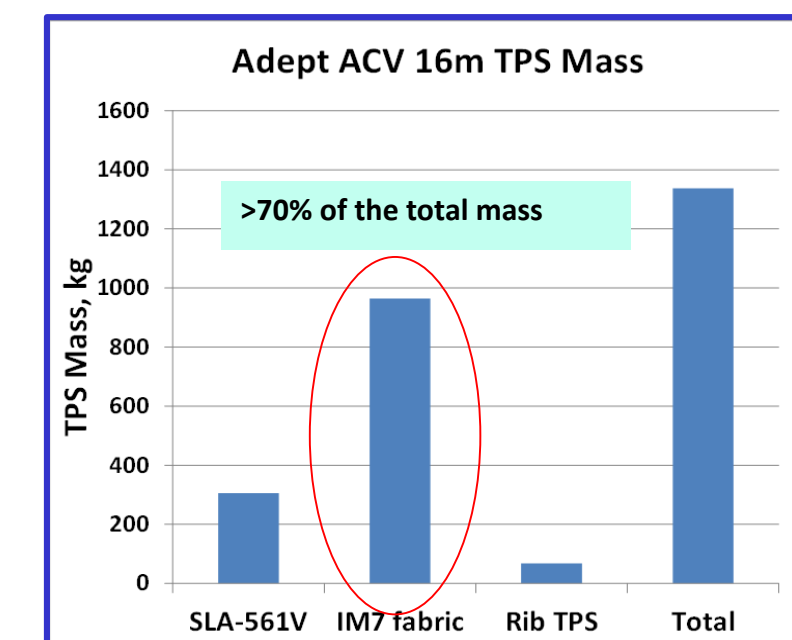
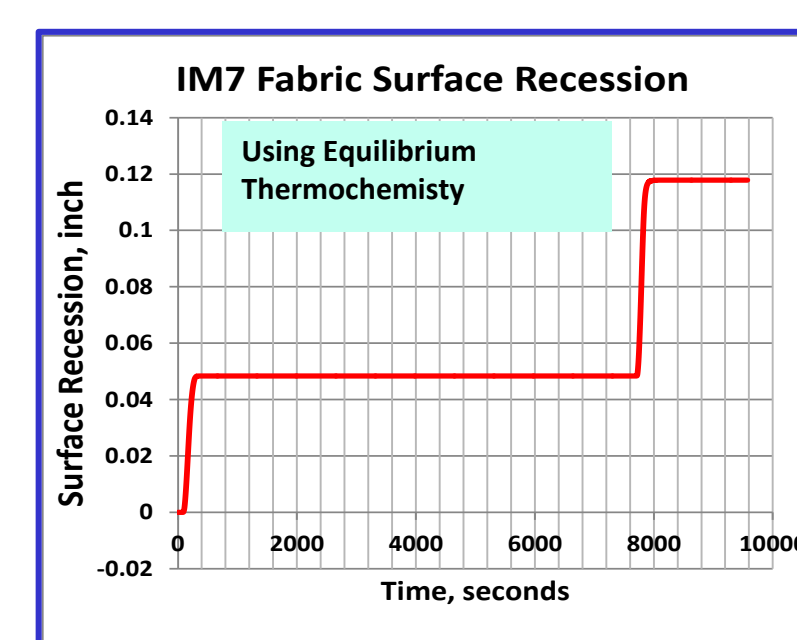
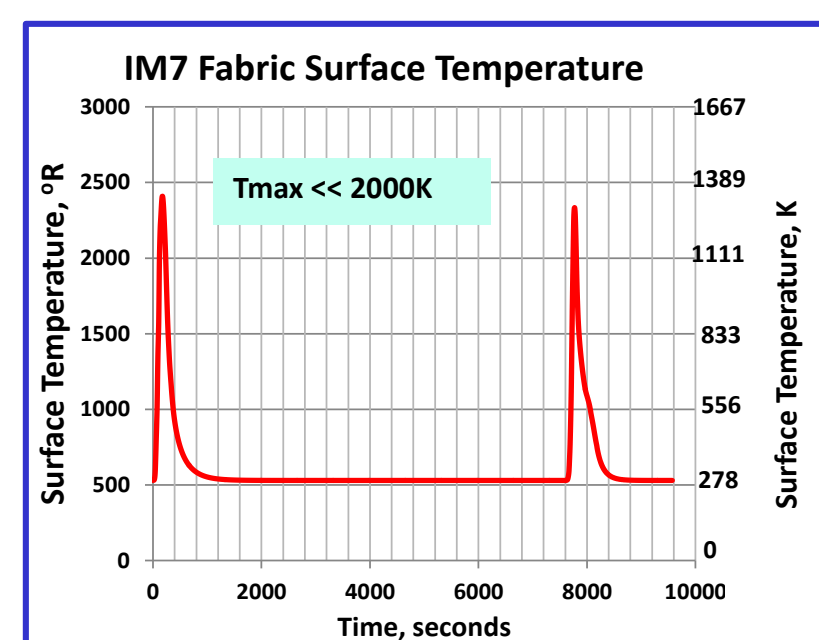
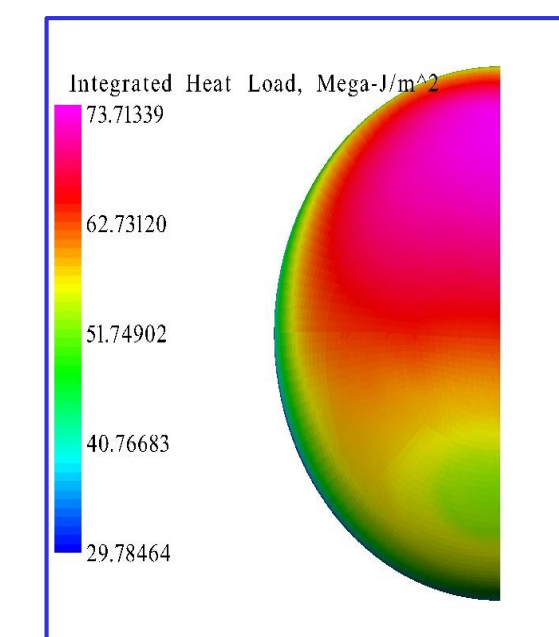
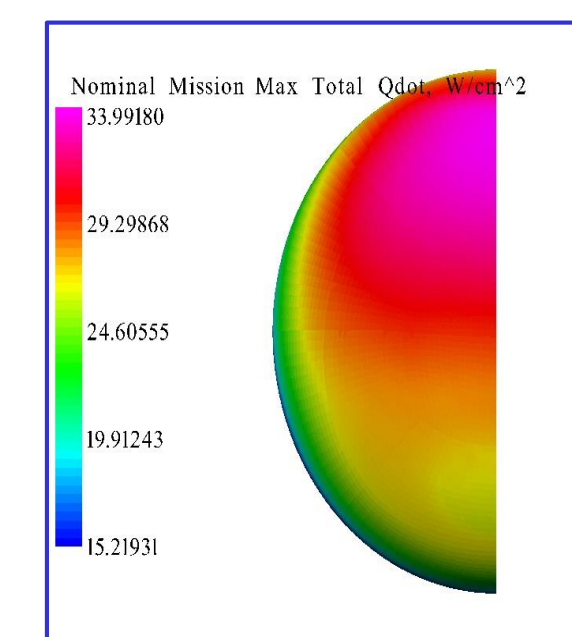
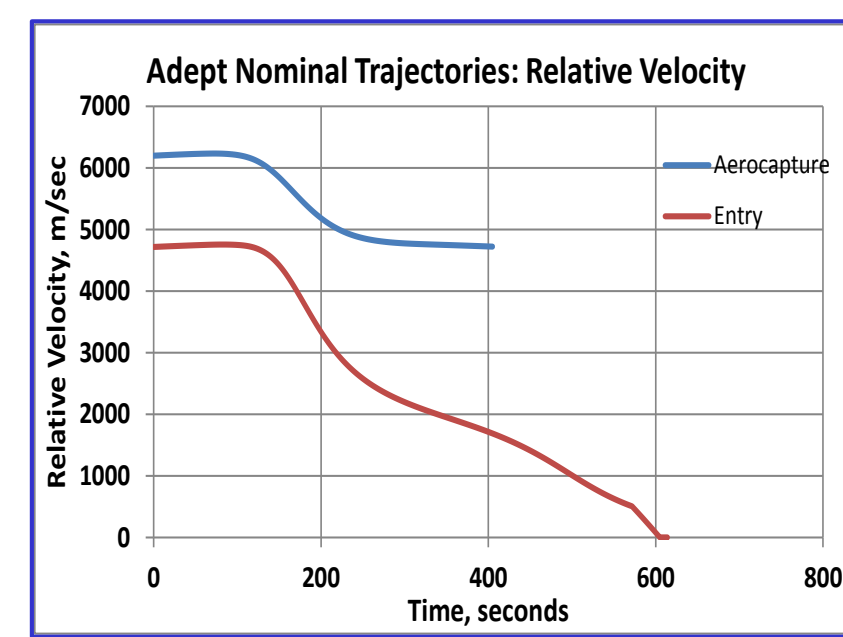
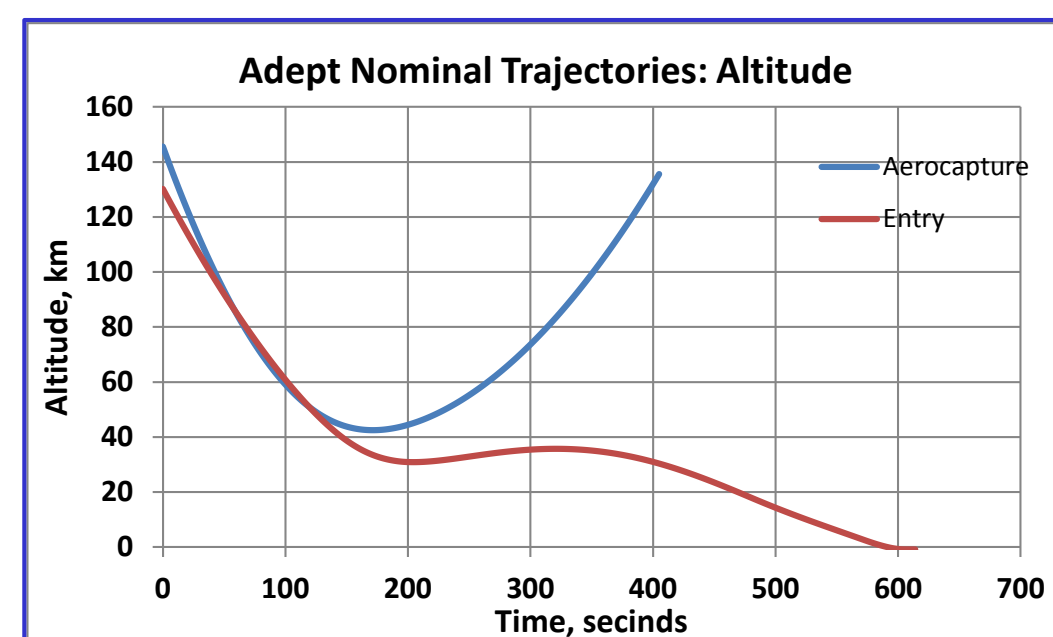
The Adaptive Deployable Entry and Placement Technology (ADEPT)

- A mechanically deployable decelerator is being considered as an entry, descent and landing (EDL) system to enable Human Mars class missions
- Ground rules for the Mars studies required aerocapture, orbit, and then entry
- Utilizes a 3-D woven carbon cloth fabric as both heatshield and primary structure
- Design guidelines required 6 layers remaining after all entry events

The Problem

- The peak heating predicted for the ADEPT carbon cloth is $<35 \text{ W/cm}^2$ and resulting temperatures were predicted to be $<1400\text{K}$
- Predictions for carbon mass loss were performed using equilibrium thermochemistry, which is only accurate for $T > 2000\text{K}$
- Carbon oxidation is kinetically controlled at $T < 2000\text{K}$, and mass loss drops off considerably from equilibrium values
- Equilibrium predictions resulted in a 15-layer carbon cloth design, with the cloth representing $\sim 70\%$ of the TPS mass
- Design of the cloth thickness and mass would be significantly reduced if kinetics were considered, but development of the kinetic constants for Carbon in CO_2 would be costly and difficult to implement in the trade studies

1: Background



Figures from a report by Jeffrey Bowles (ARC), Steven Tobin (LaRC) and Stanley Bouslog (JSC)

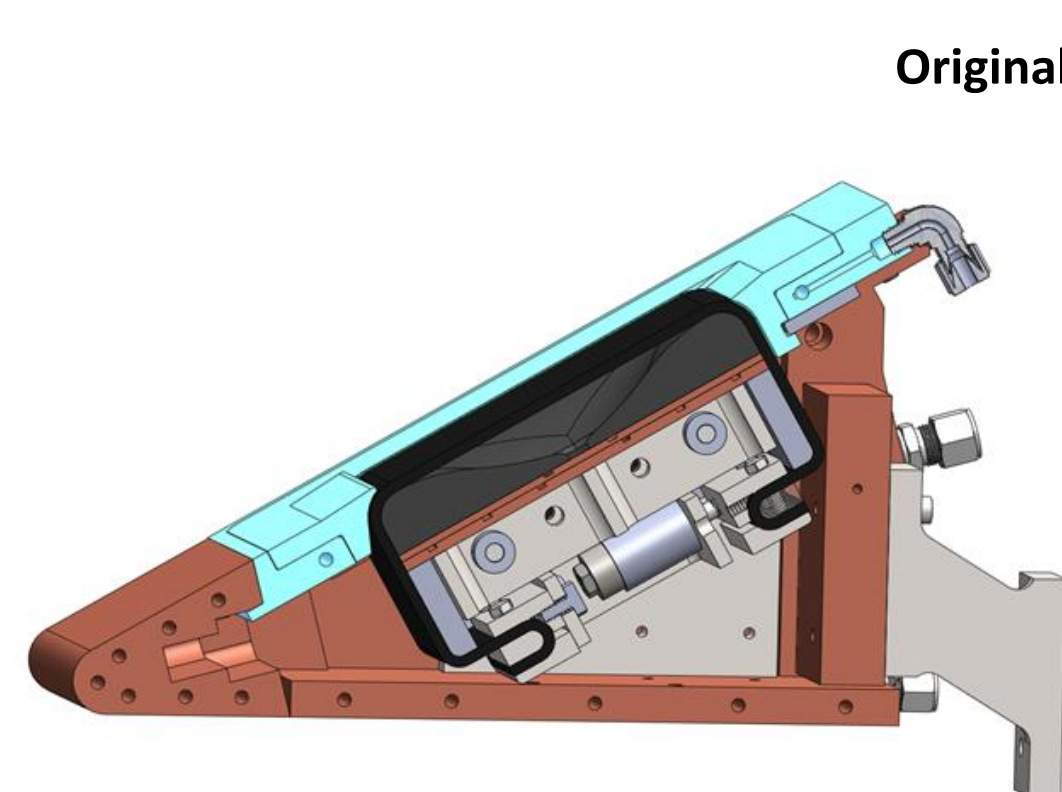
The Solution (This project in red)

- Develop an *engineering model* to describe the recession rate of the carbon as a function of the partial pressure of monatomic oxygen, which could easily be implemented in the trade study computational stream
- The AHF uses Nitrogen, Oxygen and Argon rather than Air and Argon (like the IHF) for testing
- Develop a stagnation test article design that can be used in the AHF with varying levels of Oxygen
- Develop a relationship for the recession as a function of the oxygen concentration

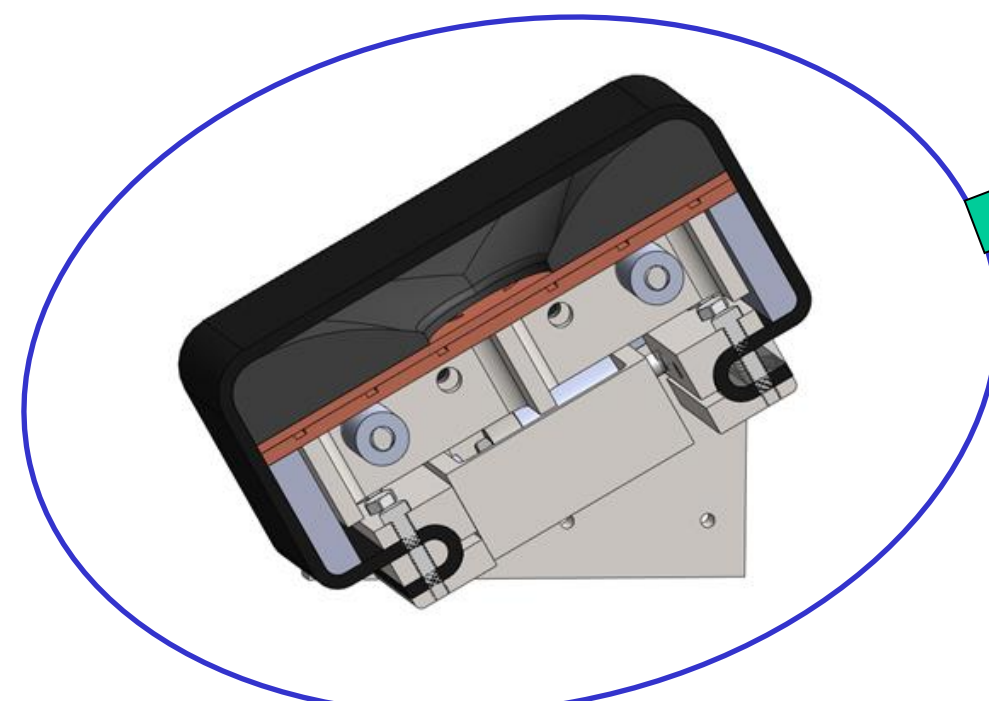
2: Design Approach

The Approach

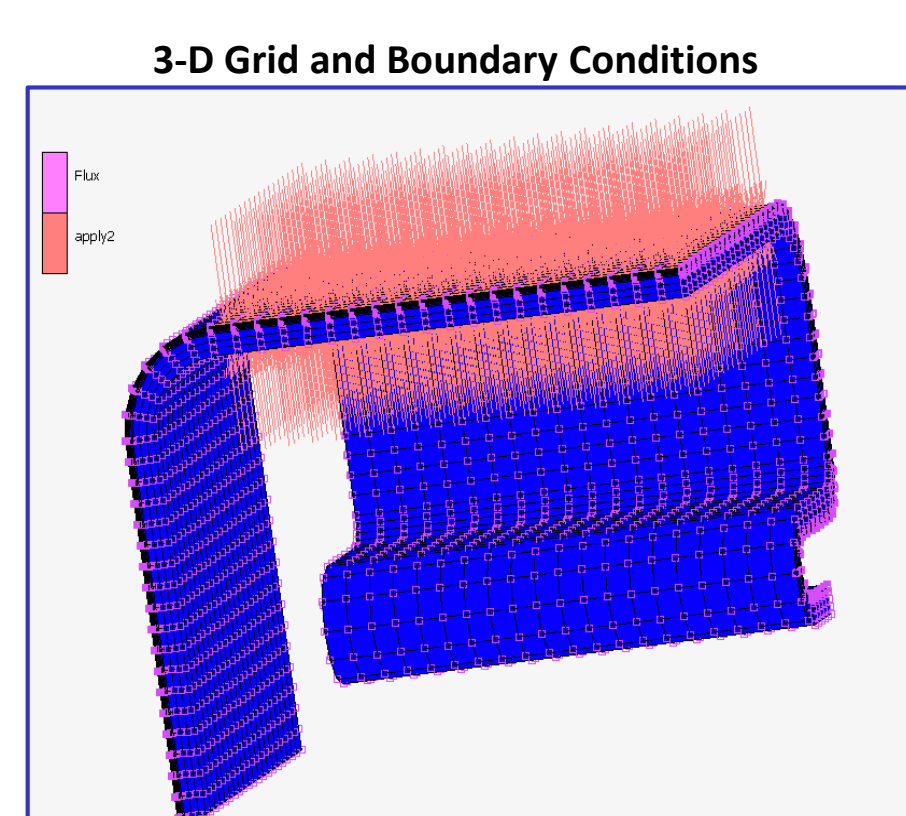
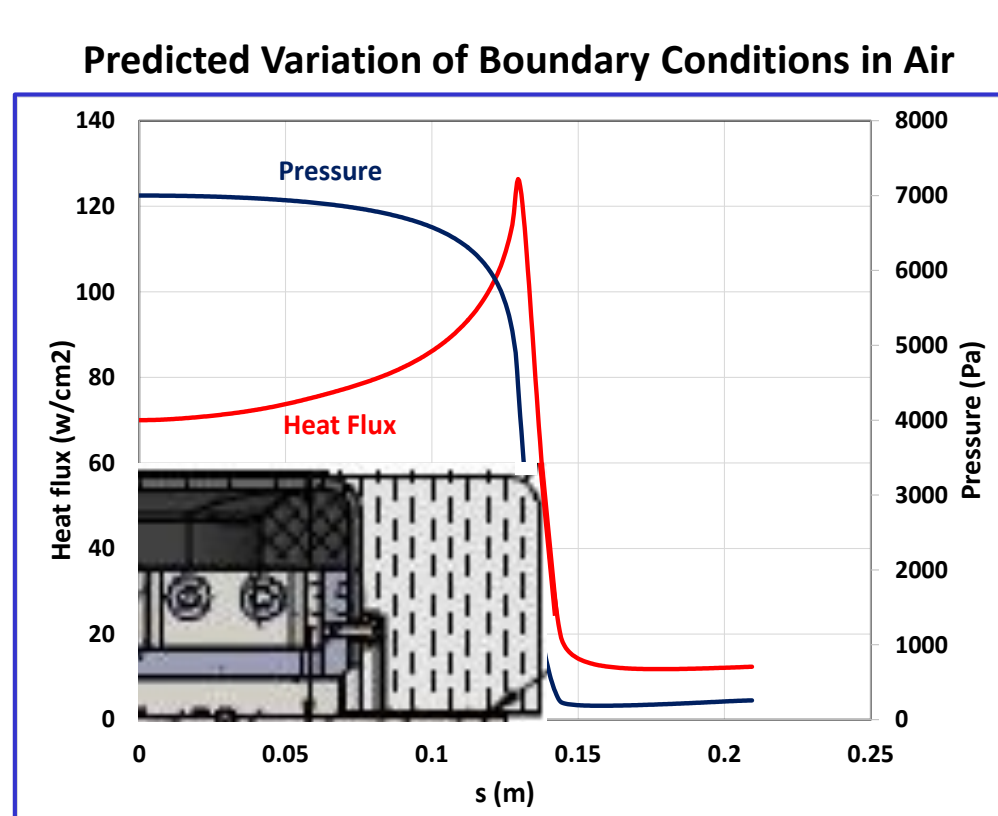
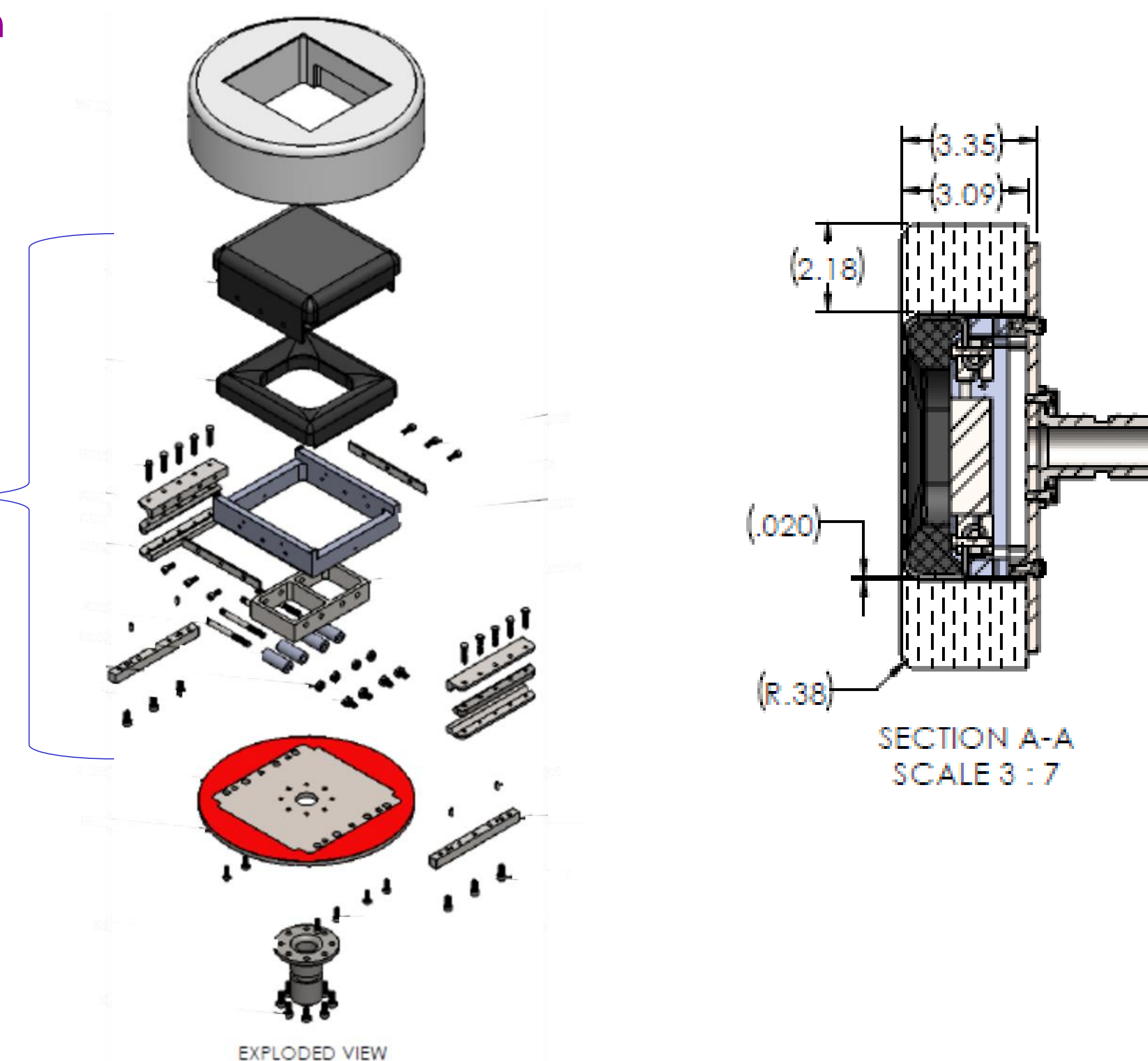
- Repurpose an existing nearly square fabric tensioning design from earlier ADEPT testing (Bilaterally Loaded Ablation Model [BLAM]) that was used in a wedge holder to evaluate the response of the cloth in shear while under load
- Simplify the design by removing the load cell and therefore negating the need for a water cooled part
- Place an insulating collar around the tensioning section and a new back plate
- Build a prototype (in process)



Original ADEPT BLAM design



Simplified, modified



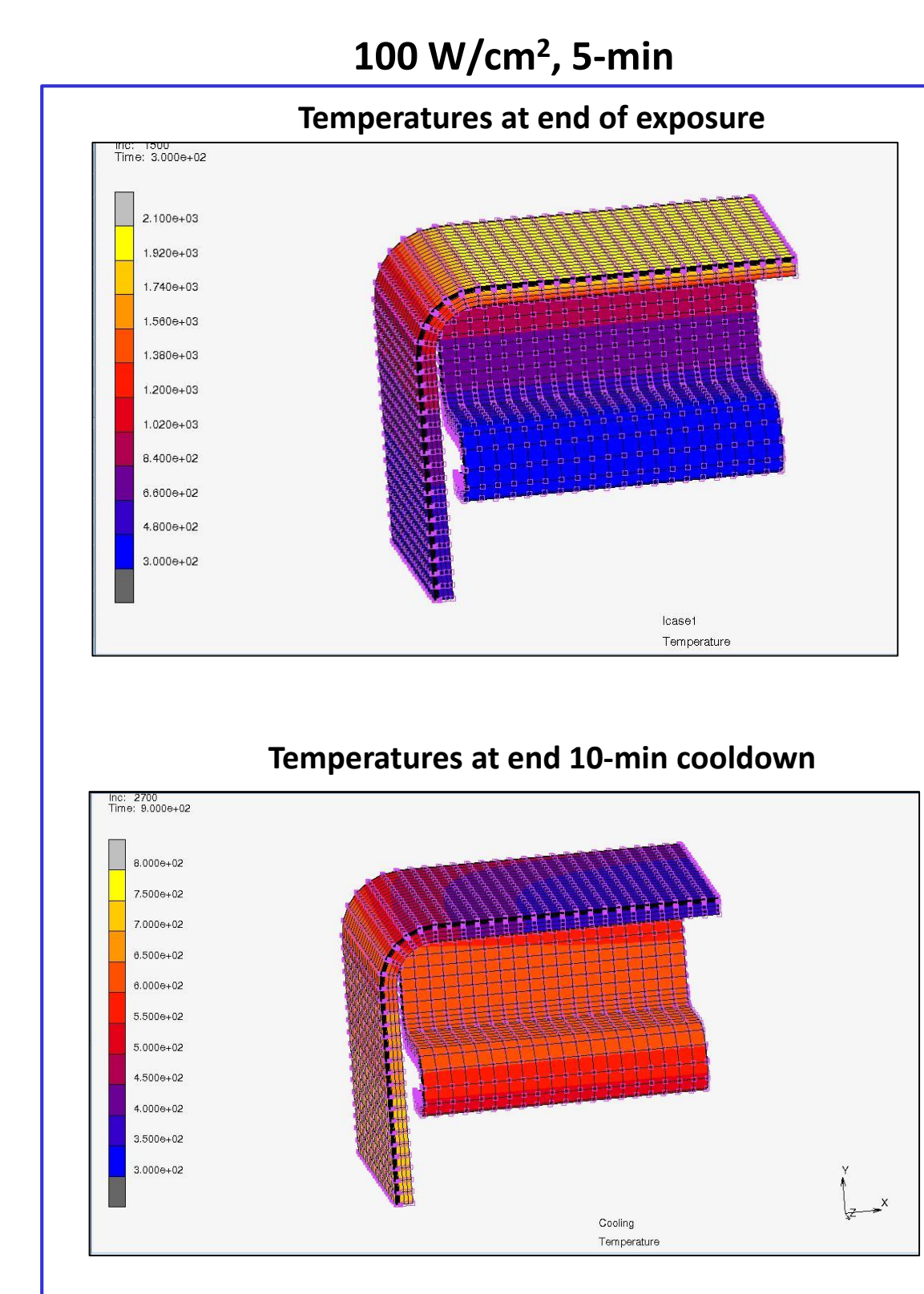
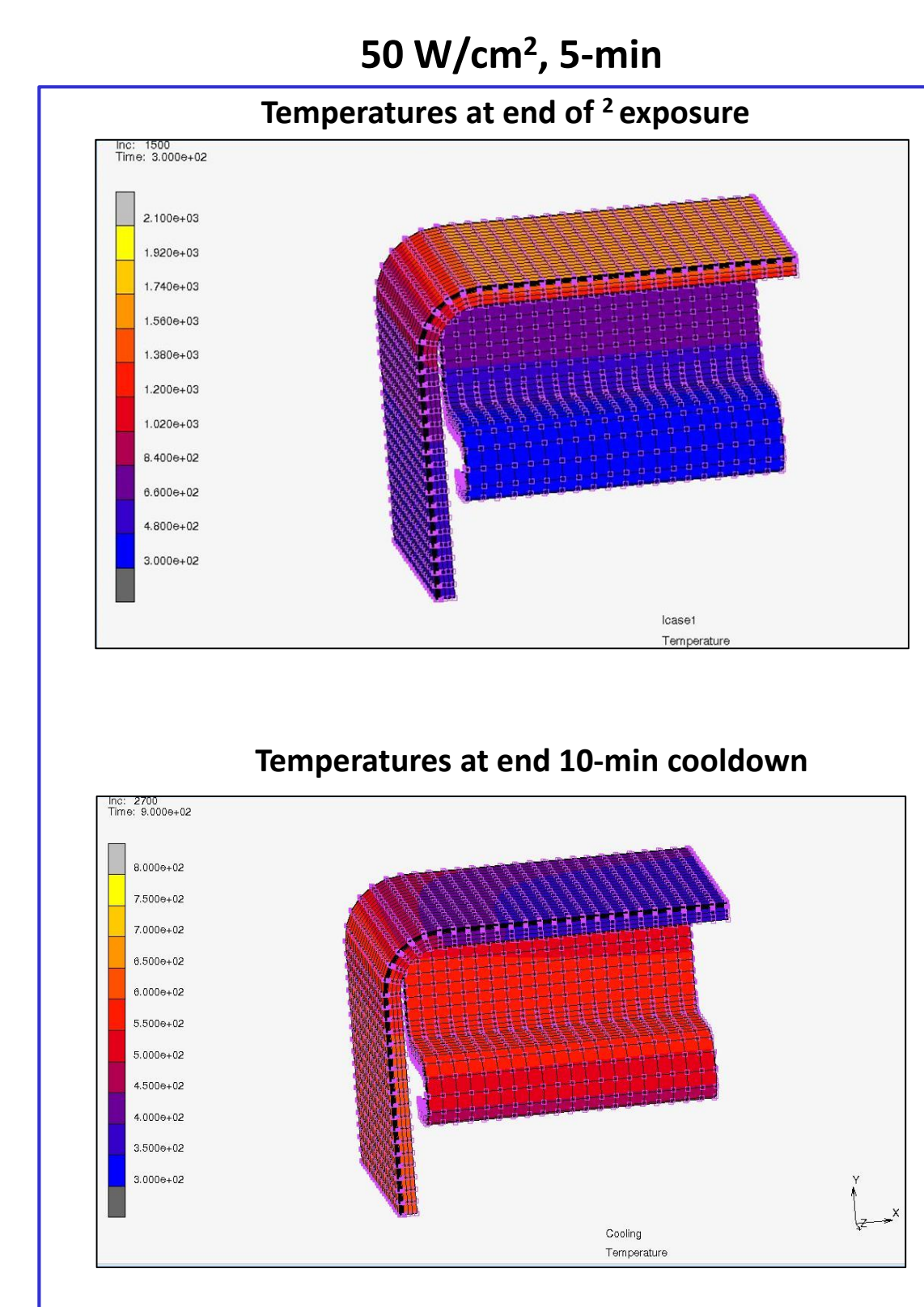
3: Analysis

Analysis Approach

- CFD analysis of a typical AHF test condition on model with a target of $\sim 70 \text{ W/cm}^2$ (2x predicted entry environments)
- Very conservative 3-D Finite Element model developed for the new carbon cloth design
 - 20,800 hex elements, 23814 nodes
 - 100 W/cm^2 and 50 W/cm^2 constant heatflux applied to top surface for 5 minutes, followed by 10 minute cooldown
 - Only top and bottom surface re-radiating to the environment, all other surfaces adiabatic
 - Transverse isotropic properties included for thermal modeling
 - Carbon cloth has much higher conductivity in-plane than through the thickness
- Future analysis work would include the Li2200 collar and the graphite frame beneath the cloth

The Results

- Analysis shows that the collar material will survive heating due to the carbon cloth in proximity ($T_{\text{carbon}} \ll T_{\text{melt, Li2200}}$), as will all other materials in contact
- This design should work well in the AHF in flows with heatfluxes at or below 100 W/cm^2 with no loss of material integrity



4: Summary

A new stagnation test article has been designed for developing an engineering model representing the mass loss of carbon cloth as a function of the partial pressure of monatomic oxygen for more reasonable predictions of carbon cloth thickness requirements in low heating environments

5: Acknowledgements

- This work was funded by NASA Ames FY17 Director's Discretionary Fund
- The Human Mars ADEPT analysis was funded by STMD GCD
- The original BLAM design was funded by STMD GCD