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Assignment 61 501 MEL R&D Phase Report 283/64 April 1965

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#### ABSTRACT

Reported here are the final design changes to a silver-cadmium battery sensing and switching circuit developed at MEL for satellite electric power systems. This circuit is designed to continuously monitor the terminal voltages of an active and a standby battery and the individual cell voltages in the active battery. When minimum voltage conditions are met, the circuit switches electrical load from the active to the standby battery. This circuit also allows solar cells, when energized, to supply load current and battery charging current.

Also described is a circuit designed to control the charge current into recently developed nickel-cadmium cells with control electrodes ("three-terminal" cells). The potential on the control electrode of a cell is the input signal to this circuit which, in turn, determines when and how the charge current is to be changed. At present this is a laboratory instrument only but will be made suitable for satellite applications in later developments.

# ADMINISTRATIVE INFORMATION

The work described in this report was accomplished under NASA Contract S 12730-G as amended, (2) 4 Feb 1963, and (3) 18 Oct 1963.

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#### REFERENCE

(a) MEL R&D Phase Report 61 501 of 17 Jan 1964

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#### SOLID-STATE CIRCUIT DEVELOPMENT

#### 1.0 INTRODUCTION

This report will describe the developments in circuit design work on two developmental tasks being performed at MEL on NASA Purchase Order S-12730-G.

The previous phase report, reference (a), describes the needs for and initial development work performed on a sensing and switching circuit for silver-cadmium batteries. This report will describe further improvements and additional work performed on this circuit to date. Also included in this report is a description of the design and development work performed on a circuit for controlling charge current in recently developed nickel-cadmium cells with control electrodes ("three-terminal" cells). A separate, more detached report in the charge circuit for the three-terminal cells will be prepared at the completion of this development. In its present form this circuit constitutes a laboratory instrument In the future it will be temperature compensated and only. otherwise made suitable for specific satellite applications.

2.0 SILVER-CADMIUM BATTERY SENSING AND SWITCHING CIRCUIT

Problem Background. The silver-cadmium battery sensing 2.1 and switching circuit is being developed for use in satellite power systems. Two silver-cadmium batteries, or packs of cells supply electrical power for the satellite during periods of darkness when the solar cells are not energized. The purpose of the circuit under development is to monitor the individual cell voltages and the overall battery voltage of each battery while one supplies electrical power and the other is on standby. When predetermined minimum conditions in the battery supplying power are reached, the circuit interchanges the roles of the two batteries. It also allows each battery to be charged during periods of sunlight when the solar cells are able to supply power to the satellite.

2.2 <u>Basic Circuitry</u>. The sensing and switching circuit for the silver-cadmium battery, reproduced here (Figure 1) with changes, appeared in reference (a). There also was a detailed description in that report. In Figure 1, the components shown with an asterick (\*) have values and/or positions that are different from the same numbered items in Figure 3 of reference (a). The numbering of components in the figures of both reports coincides so that comparison is easy.

#### 2.3 Circuit Changes.

2.3.1 Previously the gate biasing voltage dividers for Silicon Controlled Rectifier 2 (SCR<sub>2</sub>), SCR<sub>3</sub>, and SCR<sub>5</sub>, SCR<sub>6</sub> had been connected in series. It was found, however, that at low temperatures, if SCR<sub>2</sub> (or SCR<sub>5</sub>) was biased on, then there was a chance that the biasing divider for  $SCR_3$  (or  $SCR_6$ ) might be current starved due to leakage into the gate of  $SCR_2$  (or  $SCR_5$ ). Therefore, the dividers were separated and connected in parallel as shown in Figure 1. The dividers are now Resistor 8  $(R_8)$ , Sensistor 1 (TCR<sub>1</sub>),  $R_{12}$ , for SCR<sub>3</sub>;  $R_9$ , Thermistor 3 (TH<sub>3</sub>), in parallel with (||),  $R_{10}$ ,  $R_{11}$  for SCR<sub>2</sub>;  $R_{21}$ , TCR<sub>2</sub>,  $R_{25}$  for SCR<sub>6</sub>; R<sub>22</sub>, TH<sub>6</sub>, R<sub>23</sub>, R<sub>24</sub> for SCR<sub>5</sub>. The form of temperature compensation in the gate-biasing dividers for SCR3 and SCR6 were modified by the use of sensistors (TCR) instead of thermistors (TH). By incorporating these changes, the total circuit loss is increased by only 1.5 milliwatts (mw) (0.5-ma\* drain in the new gate dividers).

2.3.2 The lower cell-voltage limit requirement for battery switching has been changed by NASA since the last phase report. The previous requirement of 0.9 volt was lowered to 0.8 volt. Therefore, the cell detector diodes  $(D_1, D_2 \text{ and } D_5, D_6)$  were changed. Now  $D_1$  and  $D_5$  are 1N457's, and  $D_2$  and  $D_6$  are 1N191's. This change was in accordance with the adjustments section of Appendix B of reference (a).

The temperature compensating networks  $(R_3, R_4, || TH_1, R_5 and R_{16}, R_{17}, || TH_4, R_{18}$  of Figure 3, reference (a)) at the output of the cell detectors have been eliminated in favor of a single compensating network at the output of the drive flip-flop.

\*Abbreviations used in this text are from the GPO Style Manual, 1959, unless otherwise noted.



USN MARINE ENGINEERING LABORATORY Silver Cadmium Battery Switching Circuit

Figure 1

## Parts List For Silver-Cadmium Battery Switching Circuit

.

$\begin{array}{rcrcrc} R_1 &, & R_{14} &- 470 \ \Omega \\ R_2 &, & R_{15} &- 10 K \\ R_6 &, & R_{19} &- 100 K \\ R_7 &, & R_{20} &- 10 K \\ R_8 &, & R_{21} &- 1.7 K \\ R_9 &, & R_{22} &- 1.8 K \\ R_{10} &, & R_{23} &- 2.8 K \\ R_{11} &, & R_{24} &- 3 K \\ R_{12} &, & R_{25} &- 1.4 K \\ R_{13} &, & R_{26} &- 40 K \\ R_{27} &- & 316 K \\ R_{28} &- 100 \Omega \\ R_{29} &- & 500 \Omega \end{array}$	All Resistors 1% ½ watt
C <sub>1</sub> , C <sub>4</sub> - 50 $\mu$ f C <sub>2</sub> , C <sub>5</sub> 03 $\mu$ f C <sub>3</sub> , C <sub>6</sub> - 200 $\mu$ f C <sub>7</sub> - 82 $\mu$ f	
$L_1, L_4$ - Core, Sprague D452-20 $L_2, L_5$ - Transformer - United $L_3, L_6$ - Relay coils $S_1$ - Relay contacts Pote	U-Fl, 600 turns each winding Transformer Corp. Di-T 36 ter & Brumfield SL11DB 12 volt
TH <sub>3</sub> , TH <sub>6</sub> - Thermistor 1K 4.8% TCR <sub>1</sub> , TCR <sub>2</sub> Sensistor 6.8K; TH-	T.C. 7 - Thermistor 3K 4.8% T.C.
$D_1, D_5 - IN457$ $D_2, D_6 - IN191$ $D_3, D_7 - IN457$ $D_4, D_8 - IN92$ $D_9 - IN1117$	$SCR_1$ , $SCR_2 - 2N2324$ $SCR_3$ , $SCR_4 - 2N2324$ $SCR_5$ , $SCR_6 - 2N2324$
Q <sub>1</sub> , Q <sub>2</sub> - 2N657 Q <sub>3</sub> , 2N2905; Q <sub>4</sub> - C624 (FET) C1	rystalonics
$v_1, v_2 - 13$ cell silver cadmin	um battery

Figure 1 (Cont)

2.4 <u>Final Drive Flip-Flop Circuit</u>. A final drive flip-flop has been selected for this circuit as shown in Figure 2. This flip-flop has an output pulse of 4.4 volts, 78 msec wide approximately every 55 seconds. The pulse frequency output can be varied by changing  $R_{27}$ . Lowering  $R_{27}$  will increase the frequency of pulse occurrence. Increasing  $R_{28}$  will increase the pulse width, and increasing the divider on  $R_{29}$  will increase the pulse amplitude. The combination  $R_{30} \parallel$  TH<sub>7</sub>,  $R_{29}$ temperature compensates the flip-flop output over the operating range of -10 to +60 C. The loss in this flip-flop is approximately 12 mw for 78 msec out of each minute.

2.5 <u>Summary and Conclusions</u>. In its present condition, the silver-cadmium battery sensing and switching circuit is complete. Performance characteristics will be confirmed by a complete series of temperature tests on this circuit as a whole before the project will be considered concluded.

#### 3.0 CIRCUIT FOR CONTROLLING CHARGE CURRENT

The development of a circuit for controlling charge current in three-terminal nickel-cadmium cells is a task which was initiated after the last phase report was issued. These three-terminal cells are a recent development whose characteristics are similar to those of other nickel-cadmium cells, but in addition have a separate terminal whose potential (with respect to the negative cell terminal) is a function of the charged state of the cell. The circuit is designed to monitor the third terminal potential of each cell in a group of five while they are being charged. As the potential rises for any one of the cells in the group, the circuit begins reducing the charge current to the series group of cells so that when the third terminal potential of any cell reaches a predetermined maximum, the charge current will have been reduced to a preset trickle charge.

In its present form this circuit is to be used as a laboratory instrument. With additional detectors, it will accommodate virtually any number of cells. In further developments this circuit will be temperature compensated and otherwise made ready for satellite applications.

A detailed description of the operation of this control circuit is included in Appendix A of this report.



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The entire charge current control circuit is shown in Figure 1-A. This circuit was specifically designed to be used with three-terminal, nickel-cadmium cells manufactured by two different companies. The characteristics of the third terminal potential of the cells from these two manufacturers are not the same. For circuit-design purposes the chief difference is that the third-terminal potential of the cells from Company B does not increase at the same rate as that of the cells from Company A. This means that the circuit controlling the charge current must have two modes of operation, one each for the cells of each manufacturer. A selector switch is provided for choosing the mode of operation desired.

The circuit consists of five sections: Theory of Operation. 3.1 a power supply, a square wave driver, a detector section containing a detector for each cell, a current control, and a time delay section. The power supply converts 115 volts, 60-cycle ac to 10 volts dc  $\pm$  1 percent regulated for input and load variations. This is the bias supply for the remaining The square wave driver provides a 500 cps, 4-volt sections. (positive only) square wave for the detector. Each detector samples the third-terminal potential for its cell 500 times a second and translates this information to a d-c level. Each detector has an input impedance greater than 500 ohms and yields complete isolation between cells and between its cell and the following circuitry. The current-control section uses the d-c level from the detectors to decrease the battery charging current from full charge to a trickle charge as a cell's third-terminal potential increases. The charge current being controlled is derived from an external supply. The time delay section, when used, causes the initial decrease in charging current to be delayed by 5 minutes after a cell's third-terminal potential reaches the level which would normally cause the charge current to begin decreasing. The highest third-terminal potential of each group of cells governs the charging rate of the group.

3.2 <u>The A Mode</u>. When used in conjunction with the cells from <u>Company A</u> (selector switch in Position A), the circuit admits a full charging current into the five cells when they are in a discharged state. As any one cell nears its fully charged condition, its third-terminal potential begins to increase from zero. When this potential reaches approximately 125 millivolts (mv), the charge control section of the circuit begins to decrease the charging current into the cells. As

any cell's third-terminal potential increases above 125 mv, the charging current decreases linearly, until the current is reduced to a trickle charge when the third-terminal potential of any cell has reached 250 mv. If the thirdterminal potentials continue to increase, the circuit will remain in this trickle charge condition until the cells are manually disconnected. If all third-terminal potentials fall below 250 mv, the charge current will increase linearly and again be controlled by the highest third-terminal potential.

3.3 The B Mode. With the selector switch in the B position, the circuit will allow a full charge current into the five discharged cells. However, in this mode when the thirdterminal potential of any cell increases to 125 mv, a 5-minute time delay is initiated. For the next 5 minutes, a full charge current will continue into the cells regardless of the condition of their third-terminal potentials. At the end of 5 minutes, control of the charging current will be returned to the cell with the highest third-terminal If this potential is still 125 mv, the full potential. charge current will continue; if it is between 125 and 250 eW, the charging current will decrease to a value proportional to it; if it is above 250 mv, the charging current will immediately reduce to its trickle charge level. That is, at the end of the 5-minute time delay, the operation of the circuit in the B mode is identical to the A mode, with respect to the third-terminal potentials. In either mode of operation, the respective cells, after the charge current has been reduced to trickle charge level, should have been charged to approximately 110 percent of their capacity.

#### 4.0 SUMMARY OF REPORT

4.1 As stated previously, the silver-cadmium battery sensing and switching circuit development project will be considered concluded after final temperature tests have been completed.

4.2 The charge current control circuit for the threeterminal nickel-cadmium cells project has been completed for the laboratory instrument phase, and two prototypes have been delivered to NASA, Goddard Space Flight Center, Code 636.2. Several of these instruments have been produced by a manufacturing firm at NASA's request and are now being used in their three-terminal battery testing program at NAD,

Crane, Indiana. In later developments this curcuit will be temperature compensated and otherwise made suitable for satellite applications.

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# Appendix A

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# Circuit Description of Charge Current Control Circuit

This appendix refers to Figure 1-A and describes the stage by stage operation and adjustments in the main portions of the charge-current control circuit for three-terminal nickel-cadmium cells.

The power supply is a standard bridge rectifier; filter, and series voltage regulator. It consists of  $Q_{10}$  and  $Q_{11}$ ,  $D_9$  through  $D_{15}$ ,  $R_{23}$  through  $R_{29}$ ,  $C_8$ ,  $L_1$  and  $T_3$ . It converts 115 volts, ±10 percent 60-cycle ac to 10 volts dc ±1 percent regulated for a load variation of 100 percent.

The drive flip-flop is a standard type. It consists of  $Q_5$  and  $Q_6$ ,  $D_7$ ,  $R_{10}$  through  $R_{13}$ ,  $C_3$ , and  $C_4$ . It provides a 500-cycle, 4-volt positive square wave to the detectors as a sampling signal.

Each detector consists of  $Q_4$ ,  $D_4$  through  $D_6$ ,  $R_6$ ,  $R_8$ ,  $C_2$ , and  $T_1$ . Resistor  $R_9$  is a 47-ohm leakage resistor from the third terminal to the negative terminal of each cell as specified by NASA. These detectors have a d-c input resistance greater than 500 ohms, and because of the transformers,  $T_1$ , leave the third terminals (control electrodes) of the cells completely isolated from each other. The OR circuit formed at the output mode of the detectors effectively isolates the detector outputs from each other. If a cell is in a discharged condition, its third-terminal potential is virtually zero; thus, no dc flows through D<sub>6</sub>, R<sub>8</sub>, and the dc side of  $T_1$ . For the square wave sampling signal in the a-c side of  $T_1$ , this path appears as a high impedance. Therefore, the energy of the square wave is stored in  $C_2$ , since the path through  $D_5$  and  $C_2$  is a lower impedance. This allows  $C_2$  to maintain a d-c voltage level sufficient to bias  $Q_4$  on in a near saturation state. As the nickel-cadmium cell charges up, its third-terminal potential increases, thus allowing a low dc to flow through  $D_6$ ,  $R_8$ , and the d-c side of  $T_1$ . As this current increases (with an increasing third terminal potential), the dynamic impedance of Diode D<sub>6</sub> decreases, thus lowering its reflected impedance as seen by the square wave. As this impedance decreases, more energy is dissipated through  $T_1$  and less is stored in  $C_2$ . This will lower the d-c voltage maintained by  $C_2$  and will cause  $Q_4$  to travel through its active region toward cutoff. When  $Q_4$  is near saturation, its collector voltage is held at less than 1 volt, but as it moves through the active region toward cutoff, its collector voltage increases from less than 1 volt toward a maximum of

10 volts (the bias supply). With Diodes  $D_4$  of the five detectors connected at their cathodes, they act as an OR circuit, meaning that the highest anode voltage is the voltage which appears at the cathode connection of the diodes. Therefore, this is the d-c output level of the detector stage.

The current control stage simply inverts a changing d-cvoltage level and converts this to a changing d-c output from an external current supply. This section consists of  $Q_1$ through  $Q_1$ ,  $Q_1$ ,  $D_1$  through  $D_2$ ,  $R_1$  through  $R_5$ ,  $C_1$ ,  $SCR_1$ ,  $S_1$ , and S<sub>2</sub>. With Selector Switch  $S_1$  in the A position, the time delay section is deactivated. A rising detector output voltage (caused by a third-terminal potential rising above 125 mv) will bias the darlington pair  $Q_{12}$ ,  $Q_3$  on from cutoff toward saturation, thus lowering the collector voltage of  $Q_{12}$ ,  $Q_3$ . As the collector voltage of the darlington pair decreases, the base current of  $Q_2$  is decreased, thus reducing its collector current. A decreasing collector current in  $Q_2$  is the same as a decreasing base current in  $Q_1$ . If the base current of  $Q_1$  is reduced, its collector current is reduced proportionately, thus decreasing the charging current from the external supply into the five nickel-cadmium cells. This will be reduced to trickle charge level when any thirdterminal potential reaches 250 mv.

The time delay section is simply a unijunction-transistor timing circuit adjusted to yield an output pulse 5 minutes after being activated. It consists of  $Q_7$  through  $Q_9$ ,  $D_8$ ,  $R_{14}$  through  $R_{21}$ ,  $R_{30}$ ,  $C_5$  through  $C_7$ ,  $C_9$ ,  $T_2$ , SCR<sub>2</sub>, and SCR<sub>3</sub>. With Selector Switch S1 in the B position, the 10-volt supply is connected to the time-delay circuit, but it is not activated until SCR<sub>2</sub> has been gated "on." Also with S<sub>1</sub> in this position, the emitter of  $Q_3$  will not be connected to ground until SCR<sub>1</sub> has been gated "on." When any third-terminal potential reaches 125 mv, the detector output voltage is at a level sufficient to gate on SCR2, through the field effect transistor  $Q_0$ , thus activating the time delay circuit. So far in this B mode,  $Q_3$  has been unable to control the remaining portion of the current control section, since its emitter is floating. Therefore, the standing bias condition on  $Q_2$ maintains the charging current at full charge. Five minutes after the activation of the time delay circuit, its output pulse through  $T_2$  will gate on  $SCR_1$ , thus allowing the detector output, through  $Q_{12}$ ,  $Q_3$ , to control the value of charging

current into the five cells. The same pulse which gated on  $SCR_1$  also gated on  $SCR_3$ , enabling the energy stored in  $C_9$  to turn off  $SCR_2$  and deactivate the time delay circuit. Now the value of the charging current will depend on the detector output and its control will be the same as that described above for the A mode.

There are a number of adjustments possible in both the detector and the current control sections. The value of the third-terminal potential at which the charging current begins to decrease (or at which the time-delay circuit is activated in the case of B mode) can be varied by adjusting  $R_7$ . Increasing R7 will allow the third-terminal potential to rise to a value greater than 125 mv before the charging current begins to decrease. Decreasing R7 will have just the opposite effect. As this adjustment is made, the value of the third-terminal potential at which the charging current reaches trickle charge level is varied also, with the difference between the two potentials remaining essentially the same. Variable Resistor R5 is included as a calibration adjustment. It is generally set when the circuit is first used to account for any difference in the gain of  $Q_3$  from its design value. It is set in conjunction with the initial  $R_7$ adjustments and afterwards only if  $Q_3$  is ever replaced. Its value will be in the neighborhood of 50 kilohms. The trickle charge value can be varied from less than 1 to greater than 2 amperes by adjusting  $R_1$ . By increasing  $R_1$ , the value of the trickle charge will be decreased and vice versa. Adjustments to  $R_1$  are made only when  $S_2$  is depressed. Switch S<sub>2</sub> is a normally closed pushbutton switch. Depressing S<sub>2</sub> allows the trickle-charge current to be adjusted without any effects from the rest of the control circuitry. The value of the full charging current can be varied from 5 to 15 amperes by adjusting R<sub>2</sub>. Increasing R<sub>2</sub> will decrease the full charge current value, while decreasing R<sub>2</sub> will increase the full charge current value. When the full charge current is adjusted with R2, the current limit on the external current supply should also be set for this value. When  $R_2$  is being adjusted, it is preferable to keep  $Q_1$  in saturation. If R<sub>2</sub> is adjusted for the desired full-charge current, and as a result the collector to emitter voltage of Q1 becomes greater than 0.4 to 0.6 volt, then the output voltage of the external current supply should be reduced to return to this condition. This is done in order to minimize the

power dissipated by  $Q_1$ . The maximum power consumed by the full control circuit from the 10-volt bias source is 250 mw (at a setting of 2-ampere trickle charge with a corresponding full charge level of 15 amperes).



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Figure 1-A Charge Current Control Circuit

Parts	List	For	Charge	Current
	Control		Circuit	-

R1		100K Potentiometer	R <sub>16</sub>	-	51 Ω
R	-	5K Potentiometer	$R_{17}$	-	100 Ω
$R_2^2$	_	549Ω	$R_{1Q}^{\perp \prime}$	-	5 Meg
R <sub>4</sub>	-	1.7 Meg	$R_{10}$	-	30 Meg
R <sub>r</sub>		2 or 5 Meg Potentiometer	R	-	4.7K
R_		10K	R20	-	19.6K
R7		20K Potentiometer	$R_{22}^{21}$	-	lK
Ro	_	470Ω	R22		<b>220</b> Ω
Rg		47 $\Omega$ Specified By NASA	$R_{24}$	-	<b>620</b> Ω
$R_{10}$	-	10K	$R_{25}^{2-1}$	-	1K Potentiometer
$R_{11}$	_	47K	R <sub>26</sub>		lK
$R_{12}$		47K All Resistors	R <sub>27</sub>	-	2К
$R_1 \overline{2}$	-	10K ¼ watt	R28	-	3.9K
$R_{14}$		51Ω	R <sub>20</sub>	-	510Ω
$R_{15}$	-	100K	$R_{30}^{29}$	-	100K
$C_1$		l uf All Capacitors	$L_1$		2h Choke
$C_2$	-	10 µf 10 WVDC or	Ŧ		Chicago Standard C-232
$C_{3}^{2}$	_	0.033 µf More	$T_1$	-	1:1 Transformer (Torola
CA	_	0.033 µf	Ŧ		Sprague Electric Rll
C	_	0.033 µf	To	-	Interstage Transform
$C_{c}^{2}$		0.1 uf	2		U.T.C. Dot-36
C-7		10 µf	T <sub>2</sub>	-	Filament Transformer
Co		7000 uf - Sprague Electric	2		Knight 6-K-48 HF
Ca	منتقد	10 uf 36D1070T	Qı	_	2N2152 (Wakefield Head
Cio		0.005 uf	~1		Sink NC 421
$D_1$	_	IN4005	0	_	2N1720 (Wakefield Head
		JAN IN457	~2		Sink NC 302 M)
-2	_	IN4005	0,	_	2N1613
- 3 D.		JAN IN457	0,		2N338
-4	_	JAN IN457	~4 0⊑	_	2N1303
- 5 D.	_	TN191	0,		2N1303
-6 D-	-	IN752A (ZENER)	~6 0_	-	2N491 (UJT)
Do	_	JAN IN457 (ZENER)	$\tilde{o}_{0}^{\prime}$	_	2N494C (UJT)
-8 Do	_	IN752A (ZENER)	-8 0	_	FE-202 Amelco Semiconduca
-9 D.	_	IN752A (ZENER)	0.	-	2N335
-10 D11	_	IN457	-10	_	2N1613(Wakefield Heat
$D_{10}$	_	IN4005	~11		Sink NF 213)
-12	_	IN4005	010	_	2N338
$D_{14}$	_	IN4005	SCR.		- 2N1871 or 2N2324
$D_{1}r$	-	IN4005	SCR	L S	- 2N2324
S <sup>1</sup>	-	DPDT Switch	SCR	2 7	- 2N2324
S S	-	Spring-loaded push-		3	
-, <b>2</b>		button switch - normally	close	эđ	
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Figure 1-A (cont)