NASA - CR - 54994 WCAP - 2870



1

TEST AND ANALYSIS OF A MANIFOLD AND RE-ENTRANT TUBE ASSEMBLY

Task 3A FEASIBILITY OF A CHEMICAL POISON LOOP SYSTEM

by	GPO PRICE \$
F. C. ENGEL	CFSTI PRICE(S) \$
A. A. BISHOP	Hard copy (HC)
	Microfiche (MF)75
managed for	# 853 July 65

prepared for

ff 653 July 65

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION CONTRACT NAS 3 - 5215

WESTINGHOUSE ELECTRIC CORPORATION ATOMIC POWER DIVISIONS

NOTICE

1

This report was prepared as an account of Government sponsored work. Neither the United States, nor the National Aeronautics and Space Administration (NASA), nor any person acting on behalf of NASA:

- A.) Makes any warranty or representation, expressed or implied, with respect to the accuracy, completeness, or usefulness of the information contained in this report, or that the use of any information, apparatus, method, or process disclosed in this report may not infringe privately owned rights; or
- B.) Assumes any liabilities with respect to the use of, or for damages resulting from the use of any information, apparatus, method or process disclosed in this report.

As used above, "person acting on behalf of NASA" includes any employee or contractor of NASA, or employee of such contractor, to the extent that such employee or contractor of NASA, or employee of such contractor prepares, disseminates, or provides access to, any information pursuant to his employment or contract with NASA, or his employment with such contractor.

Requests for copies of this report should be referred to

National Aeronautics and Space Administration Office of Scientific and Technical Information Attention: AFSS-A Washington, D.C. 20546

NASA - CR - 54994 WCAP - 2870



2

TEST AND ANALYSIS OF A MANIFOLD AND RE-ENTRANT TUBE ASSEMBLY

Task 3A FEASIBILITY OF A CHEMICAL POISON LOOP SYSTEM

by F. C. ENGEL A. A. BISHOP

prepared for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION CONTRACT NAS 3 - 5215

WESTINGHOUSE ELECTRIC CORPORATION ATOMIC POWER DIVISIONS

NASA CR-54994 WCAP-2870

SUMMARY REPORT

\$

TEST AND ANALYSIS OF DESIGN OF A MANIFOLD AND REENTRANT TUBE ASSEMBLY

Task 3A

FEASIBILITY OF A CHEMICAL POISON LOOP SYSTEM

by

F. C. ENGEL A. A. BISHOP

Prepared for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

JULY 1966

CONTRACT NAS3-5215

TECHNICAL MANAGEMENT NASA LEWIS RESEARCH CENTER CLEVELAND, OHIO REACTOR APPLICATIONS BRANCH NUCLEAR SYSTEMS DIVISION M. H. KRASNER

WESTINGHOUSE ELECTRIC CORPORATION ATOMIC POWER DIVISIONS BOX 355 PITTSBURGH, PENNSYLVANIA 15230

TABLE OF CONTENTS

1

											Pa	ge No.
LIST O	F FIC	GURES	•			•			•	•	•	ii
LIST O	F TAI	BLES			•	•	•		•		•	iv
I.	ABST	FRACT	•		•		•			•		1
II.	SUM	MARY	•									2
III.	INTI	RODUCTION			•		•			•	•	24
IV.	MODI	EL DESIGN AND OPERATION		• •		•	•				3	10
	Α.	Model Design	۰				r	ç	•		•	10
	В.	Instrumentation and Calibration			s			,	,	•	•	18
		1. Flow Distribution	,	с з	,	2				,	•	18
		2. Concentration Distribution						•			3	18
v.	TES	T RESULTS AND ANALYSIS	0					,				26
	Α.	Flow Distribution Tests					•	•	•		•	26
		1. Mechanical Design Modifications	•	•	•		•	o	٠	•		26
		2. Piping Connections Modifications	•	• :	•	•	•	2		a		26
		3. Manifold Component Modifications	e	÷ 1			,	5		٥	э	26
	в.	Concentration Distribution Tests	•	•	•		•	•	0			31
APPEND	IX A	DIGITAL COMPUTER PROGRAM "MATE"	•	•	•	•			ø	•	•	47
APPEND	IX B	CAVITATION IN INLET ELBOWS		•	•	٠		a	o		•	69
APPEND	DIX C	DERIVATION AND USE OF MIXING LENGTH .		3	•	•	•	•	o	a		71
APPEND	ם אדו	REFERENCES										74

List of Figures

•4

٠.

;

.

ş

Figure No.	Title	Page No.
III - 1	Chemical Poison Loop System - Flow Diagram	5
III - 2	Manifold-Reentrant Tube Assembly	6
III - 3	Principal Poison Tube Parameters	8
IV - 1	Manifold-Reentrant Tube Assembly - Plexiglas Model	11
IV - 2	Manifold Model Flow Tests - Manifold Pressure Top Locations	12
IV - 3	Top View, Manifold-Poison Tube Assembly Plexiglas Flow Model	14
IV - 4	Bottom View, Manifold-Poison Tube Assembly Plexiglas Flow Model	15
IV - 5	Manifold Plates Prior to Assembly	16
IV - 6	CPLS Flow Model Testing Facility	17
IV - 7	Assembly of Pitot Tubes and Conductivity Probes to Bottom End of Tubes	19
IV - 8	Manometer Board	20
IV - 9	Typical Flow Record	21
IV - 10	Concentration of Pumped Solution mg/cc-NaCl/H ₂ Conductivity Probe Calibration	0 23
IV - 11	Visicorder Readout for Conductivity Probes	24
IV - 12	Typical Visicorder Trace	25
V - 1	Static Pressure Distribution in Inlet and Outlet Plenum	30
V - 2	Data Plot, 300 gpm, Time to Bottom of Tubes	33

List of Figures (Continued)

•

•

.

V -	3	Data Plot, 400 gpm, Time to Bottom of Tubes	34
v -	4	Data Plot, 175 gpm, Time to Bottom of Tubes	35
V -	5	Data Plot, 602 gpm, Time to Bottom of Tubes	36
V -	6	Data Plot, 602 gpm, Time to Top of Tubes (Inferred)	39
V -	7	Arrival Times at Minimum and Average	
		Locations	40
V -	8	Solution Injection Tests	41
V -	9	Change in Concentration vs. Time After	
		Solenoid Valve Opening	42
V -	10	Location of Concentration Change in Tubes vs.	
		Time After Solenoid Valve Constant Injection	1.1.
		Rate	44
V -	11	Extent of Completion of Concentration Change	
		in the Tubes Inside the Core vs. Time After	
		Opening of the Injection Valve	45

List of Tables

٩,

•

Table No.	Title	Page No.
V - 1	Test Results, Flow Distribution	28
V - 2	Test Results, Concentration Distribution	32
V - 3	Comparison of Time of Arrival of Injected	
	Salt Solution at the Bottom of Tubes	38

SUMMARY REPORT

TEST AND ANALYSIS OF DESIGN OF A MANIFOLD AND REENTRANT TUBE ASSEMBLY

Task 3A

FEASIBILITY OF A CHEMICAL POISON LOOP SYSTEM

by

F. C. Engel A. A. Bishop

3 2629

I. ABSTRACT

An analytical and experimental model study was performed of a manifold and re-entrant tube chemical poison distribution system which are part of a rocket reactor control loop. The system operates by changing the concentration of the poison (cadmium sulfate) in a water solution circulating through a large number (198) of parallel tubes which project from a manifold into the core. In order to meet the specifications that (1) the flow rate through all the tubes be uniform, (2) the injected chemical poison arrive at the top of the core within 0.2 second and (3) the chemical solution concentration variation throughout the core should be within $\pm 5\%$, a special manifold and re-entrant tube assembly was designed and a full size flow model was constructed of plexiglas and a series of tests performed to prove the design.

II. SUMMARY

In order to establish the feasibility of utilizing a Chemical Poison Loop System (CPLS) for the reactivity control of a Tungsten Water-Moderated Rocket Reactor (TWMR), a poison control tube and manifold system must be designed. The performance requirements of these components are based on variation in poison concentration and velocity in the many control tubes and on the time required for changes of poison concentration to occur in the core.

The design of these components was carried out by Westinghouse Atomic Power Division under contract to NASA Lewis Research Center and is reported in references 1 and 2. As part of the same contract, a full scale plexiglas model of the reference design was built for the purposes of checking its performance. Single phase flow distribution and solution injection and distribution tests were carried out. Pressure distribution was measured to evaluate velocities and concentration measurements were made to determine time delays and poison distribution.

The plexiglas model was used in conjunction with a flow system in which total flow rate could be varied. A poison injection system was included in the flow system which allowed tests of distribution during various transients.

The test results were evaluated and compared with the design requirements and the results of the analysis. At full flow the injected poison reaches the top of nearly 18% of the 198 tubes in the reference design in less than 0.20 seconds. With a steady injection rate and full flow, the poison solution concentration in the core region reaches the end of its first transient in approximately 1.9 seconds. This is less than one circuit time through the reference design control solution loop. The maximum variation in concentration among tubes during the transient exists between a tube which was already gone through the transient and one that has not yet started the transient. For normal operation solution concentration and design rates of change in reactivity, this variation is much lower than the 5% permissible. For the xenon override solution concentration, the 5% limit is exceeded during the beginning of the transient.

Tests indicate that the flow distribution among the 198 parallel tubes is satisfactorily uniform. The variations which did exist were random and not attributable to any known design characteristic or dimensional variation. The uniformity of flow distribution did not vary significantly over a flow range of 100 to 600 gpm. The mean deviation from the average over this range varied from 6.8% to 5.3%.

- 2 -

The technique used to design the manifold for uniform flow distribution was substantiated. Only minor changes in the original configuration were required to achieve a satisfactory configuration. The total pressure drop, 90 psi, measured during the test was about 12% higher than the predicted value. This is due in a large part to the insertion of a flow straightening screen added to the system.

For single phase flow it appears that the reference design as modified during the test will meet the requirements. It also appears that the design techniques used for achieving uniform flow in this rather complex system are fairly reliable. Further two phase analysis and testing would be required should it be shown later that significant quantities of undissolved gas could be developed and retained in the system.

III. INTRODUCTION

٩,

The reference method of reactivity control of the Tungsten-Water Moderated Reactor is accomplished by means of controlling the concentration of a nuclear poisoning material in a solution flowing through the reactor core. The principle of control would be to add sufficient quantities of a high cross section chemical to the system when a reduction in reactivity is required, such as at shutdown, and to remove this chemical when additional reactivity is needed, such as at startup. A feasibility study was undertaken to design and test a system capable of accomplishing these tasks. This system, the Chemical Poison Loop System (CPLS) is shown in schematic form in Figure III-1.

The brief system description provided below is included only to help enumerate the conditions to which the chemical solution and containing system will be subjected. More detailed discussions can be found in the feasibility study task reports published by WAPD. $\frac{1}{2}$

> Solution is continually flowing through the primary piping and through the in-core tubes. When poison is required, the solenoid valve in the pressurized poison reservoir is opened, and concentrated poison solution is injected into the primary flow stream until the solution cross-section reaches the desired level. When poison must be removed from the system, the solenoid valve in the ion-exchange line opens, allowing solution to flow through this system and back again to the primary lines. The poison is removed from solution in the exchanger until the solution in the in-core tubes has been reduced in cross section to the desired level. Additional components included a system pressurizer, a main-circulating pump, a loop cooler, a distribution manifold and the required temperature, pressure and flow recording and controlling devices. Nominal loop operating conditions are 600 psia pressure, 125°F manifold inlet temperature, and 600 gpm circulating flow rate.

The principal function of the control tube and distribution manifold assembly is to distribute the solution in the reactor core in a uniform manner with a minimum loop circuit time. The manifold is located in the upper gas space of the reactor, Figure III-2. At full power operation para-hydrogen at -175°F and 600 psi pressure flows through this space. Full power operation is limited to 1 to 10 hours duration, however, extended periods of operation during after-heat removal are planned. The upper portion of the 198 control tubes is also surrounded by hydrogen. The part below the bulkhead is surrounded by moderator water at an average temperature of 205°F. There is considerable heat generation in the poison solution and the tube walls due to attenuation of gamma and neutron radiation. This heat must be removed from the solution so that steady state operation can be maintained. Without auxiliary cooling only about 10% of the heat load can be transferred to the hydrogen across the walls inside the reactor pressure vessel and therefore an additional heat exchanger is required.



4

Chemical Poison Loop System Flow Diagram

Figure III.1



*

Manifold-Reentrant Tube Assembly

- 6 -

The Reference Design evolved during the study is shown in Figure III-2. In this design the single inlet pipe is connected to the distribution manifold close to the central hole in the manifold provided for passage of the re-entrant bleed fuel assembly piping. The CPLS inlet pipe feeds into an inlet distribution header designed to provide uniform flow of the poison solution to the inlet plenum. The inlet plenum distributes the poison solution radially outward to the re-entrant tube inlets. The plenum height decreases towards its outer diameter in order to maintain a sufficiently high solution velocity as the flow decreases. After passing through the re-entrant poison tubes, the poison solution passes into an outlet plenum. The height of this plenum increases towards its outer diameter, where the solution enters an outlet distribution header and is transported to the single outlet pipe. Both the outlet and the inlet header have flow-cross sections whose dimensions vary with the through-put rate in order to maintain uniform poison solution velocities.

The manifold rests on coupling tubes which are supported by the reactor inlet tube sheet. Five tie straps are provided for laterally bracing the plena and tube assembly to the reactor vessel. The manifold plates are braced internally by a series of tie rods to permit the use of a minimum plate thickness. Penetration through the pressure vessel of the inlet-outlet lines to the manifold are made through flexible couplings. The poison tubes are sized to give the same down- and up-flow areas. Principal poison tube parameters are given in Figure III-3.

The purpose of the overall study was to design a device for supplying equal flow to 198 identical re-entrant parallel tubes arranged in a specified array. One way to achieve this objective is to make certain that the pressure drop across each path is the same. The static pressure in a manifold of uniform cross section decreases along the length because the flowing mass is being reduced by out-flow to the re-entrant tubes. Thus out-flow is not uniform. To maintain the same pressure level throughout, the manifold requires a special design to keep the velocity constant or increasing to overcome the frictional pressure drop. The manifold shape selected to accomplish this criterion can be calculated using a modification of a method proposed by Acrivos, et al/3 which is simply a solution to the combined single phase continuity and momentum equations for the system. The flow is considered to be incompressible and the energy additions are not considered to influence significantly the flow distribution. The designs of the headers and the plenums were calculated. However, the theoretical contours could be approximated closely enough by linear reduction of the height of the inlet and outlet plenums and inlet header or the width of the outlet header. The design analysis discussed in greater detail in NASA CR-54420 (WCAP-2803).

The purpose of the model study discussed in this report was to demonstrate that uniform flow can be maintained in all 198 poison tubes, that the injected concentrate will reach the core within the specified time of 0.2 seconds and that a concentration maldistribution greater than \pm 5% does not exist among tubes. The four major parts of the chemical poison system design analyzed by the Thermal and Hydraulic Development Group at APD were related to the design of the plenum and were:

- 7 -

125
.000630
10
2.48 × 10 ⁶
.000630
10
2.48 × 10^{6}
68.3
600
0.0283
0.0111
52
36



Figure III-3 Principal Poison Tube Parameters

- 1. Design of a plenum and re-entrant tube assembly for distribution of chemical poison solution.
- 2. Achievement of uniform flow distributions among the tubes.
- 3. Timing and control of concentration changes.
- 4. Heat removal required to maintain desired fluid and metal temperatures.

A full scale model of the manifold and the 198 poison tubes in their hexagonal array was constructed of plexiglas for low pressure and room temperature flow distribution and solution injection tests. Plexiglas was used to obtain a visual history of chemical concentration and flow distribution during steady state and transient single and two-phase tests. The tests were conducted in a hydraulic facility at WAPD; parallel programs of evaluation of materials and total system response tests were also conducted. The purpose and scope of all the test efforts were reported on in References 1 and 2.

IV. MODEL DESIGN AND OPERATION

A. Model Design

The model was constructed using the plexiglass so that flow and concentration patterns could be made visible by gas or dye injection. The walls were designed to withstand a calculated pressure drop of 100 psi. All internal dimensions of the model were those determined by the flow distribution analysis for the prototype, except the inlet and outlet piping nipples which had a minimum throat of 2-1/8 inch I.D. rather than the 2-1/4 inch of the prototype. Some breakage of the threads of the outer tubes was experienced and metal nipples were substituted at the upper junction where the tubes were threaded into the manifold plate.

The original design of the model and its component parts are shown in Figure IV-1. Inlet and outlet piping configurations of the prototype are simulated as closely as possible using the same length of pipe and the same number of elbows between the shutoff valves and the model inlet plenum. A deflector(1) plate prevents the incoming jet from impinging into the holes in the inlet distribution plate(3). The height of the inlet distribution annulus(2) decreases in the direction of flow such that the flow velocity is approximately constant and thus the pressure head across the perforated annular distribution plate(3) is uniform along its length. The inlet plenum(4) and outlet plenum.(5) as well as the outlet distribution header(6) are tapered in order to maintain a constant velocity and a constant static head. The fluid passes from the inlet header(2) through the perforated distribution plate(3) then through the inlet plenum(4) into the down flow tubes, (7) up the poison tube annuli, (8) radially outward through the outlet plcnum, (5) and finally through the perforated outlet distribution(9) cylinder into the outlet header(6) and the outlet pipe.(10)

The manifold is fabricated using three plates. Additional spacers can be inserted to modify the plenum width if test results indicate modifications are required. Most pressure seals are made by replacable neoprene O-rings. The "poison" tubes have a plug(11) at the lower end which carries either a pitot tube(12) or a conductivity probe.(13) The stainless steel inner flow tubes(7) are held concentric with the outer tubes(14) and the pitot tubes by two levels of three set screws (15) There are static pressure taps at forty eight locations in the manifold wall, as shown in Figure IV-2 which permit flow distribution patterns to be correlated with pressure gradients in the plenums.

() Numbers refer to balloons on Figure IV-1.



MANIFOLD MODEL ASSEMBLY Figure IV-1



•

Figure IV-2

The deflection of the plenum plates is minimized by 38 tie bolts(16) located such that they limit the maximum beam span. Six of these tie bolts are drilled out for additional pressure taps. There are 98 pitot tubes and 98 conductivity probes. The plugs carrying these instruments are interchangeable so that they can be installed using any desired pattern. The model can be taken apart and different distribution plates having different performation patterns can be inserted. Gas or dye can be injected through one of the pressure taps or through one of the centering screw holes. Figure IV-3 is a top view of the manifold model installed in the test loop. Figure IV-4 shows a bottom view of the tubes with pitot tubes and conductivity cells mounted at their lower ends. Figure IV-5 shows the plates before assembly.

The test loop shown in Figure IV-6 consisted of a 1000 gallon supply tank, two 600 gpm pumps arranged in series to obtain the maximum head, and six inch schedule 10 stainless piping including a straight measuring section containing an orifice. A bypass cooler removes the pump heat and an in-line filter keeps the water clean. A flow regulating and a back pressure valve are provided; there is also instrumentation to measure temperature, pressure, and flow. The concentrate injection system contained a chemical measuring pump, a rotameter and a solenoid valve. Injection rate was set by adjusting the pump stroke.



Figure IV-3 Top View, Manifold-Poison Tube Assembly - Plexiglas Flow Model



•





۰.



•

۰.

CPL:5 Flow Model Testing Facility



- 17 -

B. Instrumentation and Calibration

1. Flow Distribution

The total circulating flow rate was measured by an .rificemanometer which was calibrated to 1% accuracy by comparison of flow rates through an orifice which had previously been calibrated at the Pennsylvania State University Hydraulics Laboratory. The flow distribution among the poison tubes was determined by pitot tubes installed into and concentrically with the inner tubes at the lower end, Figure IV-4 and Figure IV-7. These pitot tubes were calibrated in place in a typical poison tube assembly mounted in a separate low capacity (5 gpm) calibration loop.

The effects of depth of insertion and of eccentricity in installation of the pitot tubes into the low capacity loop were checked and found to be small. At a nominal flow of 3 gpm/tube, tubes with a pitot tube installed had 8% less flow than tubes without instrumentation and tubes with conductivity cells had 1% less flow than tubes without instrumentation. This flow rate variation did not introduce errors in the analysis since only flow rates in instrumented tubes were compared with one another in the normalizing process.

The pitot tubes were mounted in the bottom plugs for the poison tubes as shown in Figure IV-7. The pitot tube pressures were read off banks of water filled precision glass tubes Figure IV-8. These tubes were manifolded and pressurized in groups by regulated compressed air. This arrangement balanced the mean pitot tube pressure, so that the desired readings could be found accurately in inches of water above the balancing pressure. The balancing pressures were indicated by mercury manometers which showed the difference between a reference pressure and the manifolds. When steady flow conditions were reached the indicating tube banks were photographed to insure simultaneous readings. Figure IV-9 shows a typical record. These records were read and the results punched on IBM cards by using a Benson-Lehner scanner. A program was written for the IBM 7094 computer to process these cards and print out the normalized flow distribution in a hexagonal array corresponding to the tube array. The program is listed and the results are reproduced in Appendix A.

2. Concentration Distribution

The time for injected concentrated solution to reach the manifold was found by conductivity probes installed at the manifold inlet and outlet and at the bottom of the poison tubes. The probes, were connected to a A-C Bridge which had its output recorded on visicorders. When a concentration change occurred there was a







•



deflection of the visicorder trace. The probes consisted of two one cm square platinum plated opposed plates as shown in Figure IV-7. They were calibrated individually and also by recording step changes in the concentration circulating solution and comparing them with chemical analyses of samples taken simultaneously. In the tests NaCl was substituted for CdSO), on an equal concentration basis. Figure IV-10 is a calibration record of 14 probes which correlates deflection and concentration. Figure IV-11 shows the three visicorders and the balancing bridge cabinet. The calibrated operating range was from 3 to 4 mg/cc solution. The injection tank charge was around 180 mg/cc. Figure IV-12 is a typical visicorder trace of a test recording time and concentration changes at the inlet, outlet and 16 tubes. There is a short delay after the valve opens before the concentration change front reaches the inlet probe. At the inlet the concentration still follows the injection pump stroke, but mixing and diffusion even out the concentration front so that it appears as a nearly linear variation at the bottom of the tubes.

The minimum time for the injected poison to pass through the manifold is found from arrival at the probe in the outlet pipe.





5



Figure IV-12

Typical Visicorder Trace

V. TEST RESULTS AND ANALYSIS

A. Flow Distribution Tests

The flow distribution tests were conducted to confirm the theoretical design of the manifold re-entrant tube assembly. Of particular interest were uniformity of flow distribution to the tubes and pressure drop. No major changes in design were found necessary and the final and preliminary designs are virtually identical and validate the theoretical design basis. Flow distributions among the tubes was uniform throughout the range of flow rates tested (100 to 600 gpm) and no singular flow region existed after the deflector plate was removed.

In the first flow test configuration 1A the tubes around the inner periphery of the manifold were partially starved.* The piping connections to the model consisted of short radius plastic elbows. These caused considerable cavitation particularly at high flow rates. Because of differential expansion a large number of re-entrant tubes at the outer periphery of the manifold failed at the roots of the threads connecting them to the manifold. For these reasons modifications were made as follows:

1. Mechanical Design Modifications

The upper threaded portions of the plastic tubes were replaced by brass nipples glued to the tubes. The tie bolts were replaced by bolts of higher strength material to prevent recurrence of failure. Threaded portions of the lower end of the plastic tubes showed radial cracks due to expansion of the plug inserts (by water absorption). They were glued, re-inforced or replaced as necessary.

2. Piping Connections Modifications

To eliminate cavitation and to reduce overall pressure drop of the test system, the outlet elbows were replaced by a straight pipe section. The sharp turn plastic inlet elbows had been selected to minimize transmission of vibration to the model. These elbows were replaced by long radius schedule 10 stainless steel elbows to suppress cavitation.** A screen was inserted downstream of the elbows to flatten the flow pro-file in the nipple.

3. Manifold Component Modifications

The flow rate through the re-entrant tubes at the inner periphery of the manifold was much lower than average. This condition was corrected by:

a. removal of the inlet deflection baffle which increased the flow through the tubes below the inlet nipple. This change and the changes made in (2) above are reported as configuration 1B.

The flow distribution test data are reproduced in Appendix A.

A discussion on the cavitation appears in Appendix B.

b. increasing the height of the inlet distribution header by 3/16 inch. This increased the flow to the average value in the tubes 180° from the inlet header. The entrance pressure drop coefficient for the holes in the distribution plate had been estimated too high as a result of treating the holes as tube entrances rather than thin plate orifices. A high coefficient increases the ratio of lateral outflow to flow through the header. When the header resistance was reduced by increasing its cross section a better flow ratio was achieved. This change is reported as configuration 1C.

In Table V-1, the test results are summarized. Using the normalized data and histograms in Appendix A, an average percentage deviation from average flow has been calculated and is listed. Also for the percentage of tubes whose flow deviated by certain increments from this average are also tabulated.

It can be seen that the minor modifications effected in the original design were successful in producing nearly average flow through all the tubes. The average deviation was improved through the modification and the number of tubes with considerable deviation from the average also decreased. For the design flow, the maximum deviation from the average is about 20% with 69% of the tubes varying less than + 5% from the average. There is the likelihood that one of the tubes for which no measurement was made as much as 30% below the average. Another point to be noted is that although the average deviation is seen to increase as the flow rate decreases, the relative flow distribution remains essentially unchanged even at the lowest flow tested.

A detailed examination of the flow data indicated that the flow rate through several tubes was considerably lower or higher than average. There was, however, no discernible pattern as to the location of these tubes. It was suspected that these irregularities were caused by some peculiarity of tube installation rather than because of geometric location in the tube array. To confirm this conclusion and to rule out instrumentation abnormality, the pitot tubes from the eight "maverick" poison tubes were moved to the minor image locations for those poison tubes. In the new location, seven out of the eight showed flow very close to average. Thus, the flow maldistribution was ascribed to dissymetry or entrance effects in the poison tube installation. Later, after this program was complete, the plexiglas model top plate was removed and the poison tube inlets were examined. No correlation between the flow pattern and tube inlet configuration (such as re-entrant or recessed entry) could be found.

V-1	
Pe	
ab.	
H	l

•

۰.

TEST RESULTS FLOW DISTRIBUTION

; from Average by: in the sume set of the se	20 <.25 <.30 <.40 <.50 Appendix A) 94 95 96 96	3 93 95 96 97	97 98 99 99	97 99 99 99	98 99 99 100 3,4) 100 100 100 100 5,6	96 98 99 100 7,8	· 98 98 99 100 9 , 10	
enuage viate:	° ℃	<u>8</u>	38	9	Ъ	ήó	100	46	97	
low Der	<.15	83	83	89	88	90	94	89	94	
μ	<,10	72	74	75	75	85	85	84	87	
	< 05	94	45	51	50	5ù	69	60	60	
Average	% Deviation	.07	LO	20		. 06	C 0 °	, 06	.06	
Flow Rate	g pm	313	296	307	455	305	602	100	310	
Run	No	<u>, –</u> 1	i7	7	ω	10	Ci Li	L3	1.5	
	Configuration*	LA	LA	1B	ΤB	TC	ЪС	() 	1C	

1A with inlet baffle 1A with inlet baffle 1B without inlet baffle 1B without inlet baffle 1C inlet header 1C heightened .180" 1C heightened .180" 1C beightened .180" Configuration

*

- 28 -
The static pressures in the manifold were measured at the outer surfaces and may be slightly different at the tube inlets. There is a small circumferential variation and a smaller radial pressure difference in both inlet and outlet plenum, Figure V-1. Thus the attempt to secure uniform static pressure was compromised slightly by the use of straight-line plenum contours instead of the more complicated curved ones calculated. The total variation in static pressure in the inlet plenum is less than 0.5 psi and 0.3 psi for the outlet plenum. The minimum pressure drop across any tube is 10.5 psi and occurred in tubes at 8.5 inch radius near the inlet. The maximum pressure drop across any tube is 11.2 psi and occurred at a 15.75 inch radius 180° from the inlet. This total spread of pressure drop of approximately 6% cannot account for the low flows observed in isolated tubes.

The pressure profile through the manifold assembly was determined from the static pressure measurements at 600 gpm:

	$\Delta \mathbf{P}$
	Measured
Location	<u>in the Model</u>
Inlet Pipe (below valve) to	
Inlet Header	25.6 psi
Inlet Header to Inlet Plenum	24,0 psi
Inlet Plenum to Outlet Plenum	10.8 psi
Outlet Plenum to Outlet Header	4.3 psi
Outlet Header to Outlet Pipe	33.0 psi
Total manifold pressure drop	97-7 psi

Because of the insertion of the screen in the inlet, the manifold pressure drop was higher than previously estimated by about 13 psi. It was concluded that the edges leading to the cutlet pipe should be rounded off or flared to reduce the pressure drop. in the reference design.

The tests were performed at atmospheric temperature and pressure without heat input to the tubes. The non-uniformity in the heat absorption in the prototype will produce a variation in the poison solution density and viscosity. Based on the reference flux distribution these variations are estimated to be less than 1% and 10% respectively.



•

٠.

Figure V-1

Figure V-1

Although originally planned no tests were conducted with tubes which had dimensions significantly different than average. More important no tests were conducted with a gas phase dispersed in the liquid. The flow distribution and pressure drop would be significantly different if a considerable amount of radiolytic gas were present.

In general, the design basis of uniform flow distribution in the tubes during single phase flow was fulfilled within practical limits. Checks at locations symmetrical with the instrumented half of the core along the plane through inlet and outlet centerlines show that the same distribution could be expected in the non-instrumented half of the core. This conclusion was confirmed in the subsequent concentration distribution tests. The mean deviation is of a magnitude equal to or less than that obtained in numerous flow distribution tests of reactor models which have a much simpler geometry.

B. Concentration Distribution Tests

These tests were made to establish the time history and spacial distribution of the propagation of concentration change through the manifold. In particular, the design required that the concentration gradient in the assembly should not exceed \pm 5%. Injection of the concentrate was to be sensed at the top of the re-entrant tubes within 0.2 seconds after opening of the injection valve. The first 16 runs were calibration runs with only ten conductivity probes connected to the recorder (array #0). When the injection system performed as designed, the 48 probes were distributed in two arrays (I and II) in succession, each covering more than one half of the core and having 16 locations in common for comparison.

The loop concentration during these tests was near 3.4 mg sodium chloride per cc solution. The concentrate injection rate was about 0.17 gpm except for runs 16 and 25 for which a higher rate was used to maximize visicorder response. It was found that the time period between injection and arrival of the concentrate at the tubes was independent of the injection rate within the limits tested. Table V-2 summarizes the conditions and results of the injection tests. The tests covered circulating flow rates from 175 to 600 gpm.

Test results obtained with the same flow rates and arrays 0, I or II were combined to give maps of time delay distribution from injection valve opening to arrival of the concentration change at the bottom of the tubes; Figure V-2 for 300 gpm, Figure V-3 for 400 gpm, and Figure V-4 for 175 gpm. The 600 gpm run was made with array II only and it's results are shown in Figure V-5. An examination of these maps indicates that the arrival time at any flow rate varies principally with radius and not very much with circumferential location. Table V-2

,

•

•

TEST RESULTS CONCENTRATION DISTRIBUTION

Deflection
Avg.

				Toon			Avg. Der	Lection			
				Concen-	Inje	ction		% of			
Run No.	Date 1965	Flow	Probe Array No.	tration mg/cc	Rate gpm	Conc. mg/cc	mg/cc	Loop Conc.	To Bottom of Tubes	To Top of Tubes	To Out- let
15	7-28	175	0	3.31	0° 1.74	122	0.122	3,70	1.62	.21	4.61
16	7-28	300	0	3°31	0°425	182	0.07	2,12	0°93	° 120	2°60
17	8-6	300	H	3° 30	0°175	186	1	,	0.923	.113	2.80
18	8-6	200	H	3-30	0°175	186	8	ł	1.31	.18	2.625
19	8-9	302	i-t	3°30	0, 17	188	0.075	2.27	0.94	.184	2,625
20	8-9	395	н	3, 29	0°17	188	0°020	1,52	0°96	.0652	2.375
21	8-11	300	TT	3. 58	0°17	160	0.070	1.96	0.94	51.	2.75
22	8-11	175	TT	3, 58	0,17	160	0,120	3.35	1.63	.22	4.6
23	8-11	407	II	3.58	0.17	160	0°050	1.40	0.71	.112	1.875
54	8-12	596	ΤŢ	3,30	0.17	164	0,0225	0.68	0°5	ł	1.6
25	8-13	600	II	3,38	0.825	200	0.2155	6°40	0,48	.075	1.44

- 32 -









A comparison of results for the same tubes from two runs at the same flow rate will indicate reproducibility of results. Table V-3 summarizes this comparative data for flow rates of 175, 300, and 400 gpm. Shown on this table are readings for all tubes instrumented during more than one test and the average values of the readings in any given tube and the average percentage deviation.

The reference system requirements for time of arrival of poison in the system were checked by direct measurement of concentration change history after injection of chemical solution. The check points are located at manifold inlet and outlet, and at the bottom of the tubes. The test results in the form of visicorder traces showed the start, duration and completion of the change (Figure IV-12). The local rate of change depends on the width of the mixing zone and its change with passage through the manifold. An experimental equation by F. Sjenitzer/<u>4</u> is derived in Appendix C. It describes the width of such a mixing zone.

s = 1.96 x 10⁴ $(\frac{L}{a})^{0.07} \sqrt{aL} x \gamma^{1.8} erf^{-1} (2y-1)$

This gives a mixing zone of about 1.3 ft length or of 0.122 seconds for the 4-1/2 ft long poison tubes and about 1 ft or 0.03 seconds for the approach piping.

The time interval required for flow to travel from the top to the bottom of the tubes at the average velocity and the time equivalent of the change in width of the mixing zone were subtracted from the measured arrival times at the bottom to give, for 600 gpm, the distribution of concentration change arrival times at the top of the tubes after opening of the injection Valve, Figure V-6. It is shown that at about 18% of the tube locations the concentrate will arrive within 0.2 seconds. Figure V-7 shows the arrival times at minimum and average locations for 600 gpm (Run #25).

The time schedule is, of course, a function of flow rate. The greater the flow rate, the shorter the times. Time schedules resulting from all the tests were prepared in the manner indicated above and the results are plotted in Figure V-8. The slope of these lines indicates that the time intervals are inversely proportional to flow rate.

The visicorder traces indicate that the change in concentration is linear with time. The duration of the change, i.e. the width of the mixing zone, increases with distance from the injection point, i.e. with time. This is illustrated in Figure V-9 where the times to start and complete the concentration changes at manifold inlet, manifold outlet, and bottom of tubes are plotted in proportion to their appearance in the test records. The slope of the top-of-tube times were calculated using the expression of Sjenitzer for the width of the mixing zone. Table V-3

•

Comparison of Times for Arrival of Injected Salt Solution At the Bottom of Tubes

Flow Rate		175	gpm			300	gpm			400	gpm	
Test Run No.	15	22			19	21			20	23		
Tube			Aver: Deviat	uge Ston			Aver: Deviat	age tion			Avera Devia	age cion
Location	Sec.	Sec.	Sec.	,0, +	Sec.	Sec.	Sec.	₽ +	Sec.	Sec.	Sec.	% + !
0-21	2.75	2.43	2.59	6.2								
C-11	х. -	<u>)</u>			1.44	1. 63	1.535	6.2	1.19	1.20	1.195	0.4
G-11					1.00	1,12	1.06	5.6	0.78	0.71	0.745	4.7
J-11	1.62	1.63	1.625	0.3	0, 94	0°94	0,94	0.0	0.71	0.73	0.72	1.4-
0-11	1.81	1.82	1.815	0,3	1.10	1.13	1.115	1.3	0.81	0.74	0.775	4.6
S-11					1.25	1.31	1,28	s. S	0.84	0.96	0.90	4.5
W-11	2.75	2.70	2.725	0.9	1.60	1.63	l,615	0.9	1.20	1.38	1.29	7.0
A-10	2.45	3.03	2.74	10.6	1.90	1.90	1.90	0.0	1.40	1.45	1.425	1.8
Н-16	1.82	1.88	1.85	1.6	1.15	1.18	1.165	1.3	0.83	0.75	0.79	5.0
I-10					0.93	1.06	0.995	0.0	0.71	0.66	0.685	3.7
P-10					1.10	1.20	1.15	4.4	0.82	0.78	0.80	2. 2
X-10					1.80	1.78	1.79	0.6	1.34	1.40	1.37	2. 2
Q-6					1, 37	1.33	1. 35	1.5	0.95	1.00	0.975	2.6
с - С					1.77	1. 78	1.775	0.3	1.13	1.25	1.19	5.0
U-4					1.75	1.75	1.75	0.0	1.35	1.33	1.3 ⁴	0.7
H-2		1			1,85	1.78	1.815	1.9	1.16	1.34	1.25	7.2
0-1	3.12	3.08	3.10	0.7	1.95	1.95	1.95	0.0	1.38	1.46	1.42	2•8
Avg.				3.0				1.8				3.5

- 38 -



•



VALUE AT POINT A', B', ETC. REPRESENT TIMES TO A TUBE AT THE AVERAGE RADIUS. VALUES AT POINTS A, B, ETC, REPRESENT MINIMUM TIMES. *



:

Figure V-8 - 41 -



•

Figure V-9

- 42 -

Figure V-9

Curve V-9 was cross plotted to show in Figure V-10 and V-11 the concentration distribution in the tubes at any time between injection and completion of one circuit after injection. Figure V-11 was obtained by defining three groups of tubes located centrally, average radius and peripherally and calculating the progress of concentration change rather than using the times for each individual tube. The percentage of the total poison tube length of the core that has undergone the concentration change is also shown, as a dotted curve. The normalized tube length completing the change can be translated into core volume fraction.

The time required for completion of one pass through the CPLS Reference System was calculated for 600 gpm from knowledge of the volumes, shapes, and mean velocities in the components:

	Volume Gallons	Length Ft。	Mean Velocity <u>Ft/sec</u>	Mean Residence Time Seconds	Approx. Pres. Drop psi
Piping, including pump and valves	3∘6	12	23.8	0.366	
Distribution manifold	6_92	-	2.5	1.666*	90
Poison tubes	7.84	4.5	10.7		
Heat exchanger	4.2	3	14	0.214	50
Total Circuit	22.56			2.246	140
Check	22.56 x	60 sec/m	in ÷ 600 gp	m = 2.26 sec.	

*From measurements.

Figure V-9 shows that it takes 2.1 seconds for a fixed concentration change throughout the manifold to be completed. If the injection valve remains open a further increase in concentration takes place as the mixing front completes one circuit. But since the total circuit time (about 2.3 seconds) is longer than the time for completion of the concentration change in the manifold, the maximum difference in concentration between tubes that can exist is that at the beginning, when some tubes have completed the first change while others have not yet started to be reached by the concentration increase. Table V-2 shows that for 600 gpm and an injection rate near 0.212 gpm this maximum difference is well below 5% of the loop concentration (Runs 2¹/₄ and 25); in Run 25 the injection rate was four times the nominal one. This produced a quadruple change in the loop concentration and the visicorder deflection.



Figure V-10

- 44 -



:

Figure V-11

Normal concentration change for 0.212 gpm injection rate would have been less than $6.40\% \div 4 = 1.6\%$.

In summary the solution injection tests demonstrated that:

•

- a. At nominal flow (600 gpm) the injected concentrate will arrive at the top of about 18% of the number of tubes in 0.2 seconds or less (Figure V-6).
- b. Within the range tested the time of arrival is independent of injection rate and injection concentration (Table V-2).
- c. The time for the flow to pass through the manifoldre-entrant tube assembly is inversely proportional to the rate of flow (Figure V-8).
- d. Within the specified rate and concentration ranges of injection the maximum concentration gradient is well below 5% (Table V-2).

APPENDIX A

DIGITAL COMPUTER PROGRAM "MATE"

1. General

The manifold test data reduction code was written to expedite test data conversion to a solution flow distribution array corresponding to the poison tube pattern. The code was also devised to normalize the array and to plot a histogram describing the number of tubes having normalized flow of various magnitudes.

The procedure consisted of taking photographs of the manometer boards and reading the photos by the Benson-Lehner scanner which converted the readings to integers from 1 to 999 punched on IBM cards in a specified array.

The code uses calibration data to convert the punched cards into flow rates in the tubes.

```
2. Algorithms
```

V(K) = SQRF ((Press (J-1)-Press (J)) * 0.060* 64.35* C/RHO)* .670

where V(K) = Flow velocity in tube K

- K = Tube Number
- J = 2 * K
- Flow(K) = Flow in tube K, in gpm

Press(J) = Static pressure at pitot tube K

Press(J-1) = Total pressure of pitot tube K

C' = Flow coefficient - Temperature dependent input

RHO = Water density - Temperature dependent input

64.35 = gc'

0.060 = Scale factor for Benson-Lehner scanner

0.0670 = Flow profile factor (avg/max)

VNorm(K) = V(K)/(SUM V/SUM I)

- 47 -

where:

•

•

VNorm(K)	=	Norma	lized flow in tube K
V(K)	=	Flow 7	velocity in tube K
SUM V	=	Sum of	f velocities in tubes 0 to K
SUM I	=	Numbe: from	r of tubes from 0 to K whose flow rate differs zero.
3. <u>Inpu</u>	t Cai	rds	
Card	Туре	e 1	Title card, any text in locations 7 to 71 1 in location 72 if title is complete 0 in location 72 if more title cards follow.
Card	Туре	e 2	Format (213, 6X, F7.2, 8X, A4, 7X, 313) Configuration - two digit integer Temperature, ^{OF} - two digit integer > 64, < 81 Flow, GPM, four digit fixed point No. Run Number, four digit integer Date: Month, Day, Year - two digits integer each
Card	Туре	€ 3	Format (F5.1, 9F6.1) 40 cards with ten three digit integers each (16.7 times the manometer reading)

.

- - - -

ND1-5 F 130 A/O 40%-51ANDARD US1-STANDARD 46%-5TA40A80 UD4-STANDARD 03 V 01. V 1 S - M 01 C⊱volle11210v30 PAGE 5/25/65 51 111 121 131 131 231 3 11 יד ע: 4 57 2 > 25 5.7 30 ¢ 2 21 $\hat{\mathbf{u}}$ 5 17 ю. П o ۲ 61 22 24 27 9 10 0 31 3 - N 4 ŝ 25 000 AATE 4.4 T G 1ATE V(198),DELP(198),SUMV(199),MATE) MATE MATE MATE MATE MATE MATE NATE MATE MATE MATE MATE MATE NATE MATE MATE MATE MATE ATE NATC BIAN. モレイド *1TE HATE 4ATE 2416 ATE 3116 MATE MATE MATE MATE NATE 217× C=144X.03672 FOR 63F,FIG77,ASME.FLUID METERS,RHU,KEENAN+K.MaTE AATE 2175 GO TO (31,32,33,34,35,36,37,38,39,40,41,42,43,44,45),KTE^yPHATE NATE F5.1,3X,9H INFORMATION UN CONFIGURATIONS EIC. A MANIFOLD TEST DATA REDUCTION PROSRAM. F.C. ENSEL, WAPD. Ħ MANIFULD DATA REDUCTION, F.C.ENGEL, APD) TUBE VELOCITY DISTRIBUTION, NURMALIZED) = I3,3X,8H FLOW TUBE VELOCITY DISTRIBUTION, FT/SEC) 5,1010,NCON,NTEMP,FLUW,NRUM,NWU,WDAY,NYEAR RANGE) , VW(198) (28H0 TEMPERATURE IS OUT OF (2X, 8H NCON = 12, 3X, 8HNTEMP[213,6X,F7.2,8X,A4,7X,313] VNORM(198) , L(24) P(396), PRESHG(5) 5,1020, (P(J),J=1,396) IF(NTEMP-65)20,22,25 IF(ICHK) 11,11,16 ABOVE READS IN ADDITIUNAL INRUN = 44,3X,6H DATE 313) KTEMP=UTEMP-65 []] FORMAT(F5.1, 9F6.1 WUT 6, 1035 RH0=62.305 5,1045, ICHK RH0=62.289 RHU=62.297 RH0=62.281 6,1045,ICHK GO TO 339 C=5.2877 GU TO 50 GO TO 50 c = 5.2819C = 5.276260 TO 50 FURMAT (40H1 (39H (35H FORMAT (72H0 (71H (IHI) 6,1055 6,1050 FURMAT (72H DIMENSION CONTINUE FURMAT FURMAT FURMAT FORMAT FURMAT FORMAT FORMAT RIPT RIPT RIPT MUT WUT WOT MATE, 1045 1000 1035 1040 10 15 1001 1002 1005 1010 1020 16 25 1050 1055 18 20 22 32 33 31 49

•

۰,

PAGE 3

۰,

5/25/65 891 106 931 166 951 87 88 69 94 6.6 96. 98 MATE1191 83 84 85 86 80 81 82 MATE 90 16 92 6 WATE110 MATE 89 MATE89 HATE112 MATE113 WATE114 HATEILS MATE119 MATE120 VATE122 MATE123 MATE126 MATE111 MATE121 **WATE124** MATE125 MATE127 MATE128 4ATE129 MATE MATË MATE MATE MATE MATE MATE MATE MATE MATW A MANIFOLD TEST DATA REDUCTION PRUSRAM. F.C. ENGEL, WAPD. V(K) = SQRTF(DELP(K) *64.35*C/RHU)*0.670 V(K)=SQRTF(DELP(M)*64.35*C/RHU) #0.670 DELP(K)=(P(J-1)-P(J))*0.060 DELP(M) = (P(J-1)-P(J))*0.060[F(P(J-1))125,125,130 IF(P(J-1))155,155,160 SECOND CONFIGURATION VNURM(K) = V(K)/VAVGIF(P(J))125,125,122 IF(P(J),155,155,152 THIRD CUNFIGURATION IF(V(K))256,256,255 [F(V(K))262,262,270 VAVG = SUMV/SUMI DO 275 K=1,198 K=1,198 00 170 M=1,198 SUMV = SUMV + V(K)VNURM(K)=0.0 GU TO 145 60 10 249 00 TO 170 GU TO 249 60 T0.249 CO TO 275 GU TO 295 Ύ(K)=0.0 CONTINUE V(K) = 0.0CONTINUE CONTINUE CUNTINUE CUNTINUE CONTINUE CUNTINUE CONTINUE 00 256 K=2#M-1 0=(r) SUMV=0 I=IWNS J=2*M [+]=] 0 **=** I MATE, 120 125 145 160 130 150 151 25 155 170 200 202 249 250 251 ζĉζ 256 260 202 270 275 295 :

- 50 -

WOT 6,	1055	MATE6291	NUN-STANDARD
WUT 0,	1002	MATE1292	NUN-STANDARD
WUT 6,	1002	MATE1293	NUN-STANDARD
WOT 6,	1001	MATE130	NON-STANDARD
WUT 6,	1002	MATE1301	NON-STANDARD
FORMAT	(39X,2F6.2,4X,2F6.2,37X,3H_21)	MATE1310	
FURMAT	(31X,2F6.2,4X,2F6.2,4X,2F6.2,29X,3H 20)	MATE1311	
FORMAT	<pre>(23X, 2F6.2, 4X, 2F6.2, 4X, 2F6.2, 4X, 2F6.2, 21X, 3H 19)</pre>	MATE1312	
FORMAT	(15X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,13X,3H 18)	MATE1313	
FURMAT	(13X, F6.2, 4X, 2F6.2, 4X, 2F6.2, 4X, 2F6.2, 4X, 2F6.2, 4X, 2F6.2, 1X	,MATE1314	
	3H 17) .	MATE1315	
FORMAT	(15X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,13X,3H 16)	MATE1316	
FORMAT	(7X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F5.2,5X)	,MATE1317	
•	3H 15)	MATE1318	
FURMAT	<pre>(5X,F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,</pre>	MATE1319	
	F6.2,3X,3H 14)	MATE1320	
FORMAT	(7X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,5X)	,MATE1321	
	3H 13)	MATE1322	
FURMAT	<pre>(5X,F6.2,4X,2Fb.2,4X,2F6.2,20X,2F6.2,4X,2F6.2,4X,F5.2,3X,</pre>	MATE1323	
	3H 12)	MATE1324	
FORMAT	(7X,2F6.2,4X,2F6.2,4H * F5.2,16X,F6.2,4X,2F6.2,4X,2F6.2,	02513TFM	
	8H ** 11)	MATE1320	
FORMAT	[5X, F6.2, 4X, 2F6.2, 4X, 2F6.2, 20X, 2F6.2, 4X, 2F6.2, 4X, F6.2, 3X,	MATE1327	
		MATE1328	
FUKMAI	【【X,ZF6、Z,4X,ZF6、Z,4X,ZF6、Z,4K,ZF6、Z,4X,ZF6、Z,4X,ZF6;Z,5X 34 041	•MALE1329 Netelood	
FUKMAI	{	MATEL 331 WATEL 200	
EODWAT	13.513/1311 301 134 354 3 AY	NATELUJC NATELUJC	
	1. × / EI U+ EI T× / EI U+ E	WATE1354	
FURMAT	(15X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,13X,3H 06)	MATE1335	
FURMAT	(13X, F6.2, 4X, 2F6.2, 4X, 2F6.2, 4X, 2F6.2, 4X, 2F6.2, 4X, 2F6.2, 4X, F6.2, 11X	,MATE1336	
	· 3H 05)	44TE1331	
FURMAT	(15X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,13X,3H 04)	MATE1336	
FORMAT	(23X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,4X,2F6.2,2LX,3H 03)	MATE1359	
FURMAT	(31X,2F6.2,4X,2F6.2,4X,2F6.2,29X,3H 02)	MATE1340	
FORMAT	(39X,2F6.2,4X,2F6.2,37X,3H 01)	MATE1341	
FURMAT	(100H0 A B C D E F G H I J K L	MATE135	
Z E	OP QR ST UV WX)	4ATE136	
WOT 6,	301,(V(K),K=1,4)	ALE140	NON-STANDARD
WUT 6.	302, (V(K),K=5,10)	MATE141	NON-STANUARD
WUT 6,	303, (V(K), K=11,18)	MATE142	NON-STANDARD
WUT 6,	304, (V(K),K=19,28)	4ATE143	UNACNAT S-NUN

:

- 51 -

MATE,	A MANIF	OLD TEST DATA REDUCTION PROSRAM. F.C. ENGEL, WAPD.	5/25/65	PAGE 5
	WUT 6,	305, (V(K),K=29,38)	MATE144	NON-STANDARD
	WOT 6,	306,(V(K),K=39,48)	MATE145	NON-STANUARD
	WÜT 6,	307, [V[K],K=49,60]	MATE146	NON-STANDARD
	WUT 6,	308,(V(K),K=61,72)	MATE147	NON-STANDARD
	WUT 6,	309, (V(K),K=73,84)	MATE143	NON-STANDARD
	WUT 6,	310,[V(K),K=85,94)	MATE149	NON-STANDARD
	WUT 6,	311, (V(K),K=95,104)	MATE150	NON-STANDARD
	WOT 6,	312,(V(K),K=105,114)	MATEI51	NON-STANDARD
	WUT 6,	313, (V(K),K=115,126)	MATE152	NON-STANDARD
	WUT 6,	314, (V(K),K=127,138)	MATE153	NON-STANDARU
	WUT 6.	315, (V(K),K=139,150)	MATE154	NON-S TANDARD
	WUT 6,	316,(V(K),K=151,160)	MATE155	NON-STANDARD
	WUT 0,	317,(V(K),K=161,17J)	MATE156	NUN-STANDARD
	WUT 6,	318, (V(K1,K=171,180)	MATE157	NON-STANDARD
	WOT 6,	319, (V(K),K=181,168)	MATË158	NUN-STANDARD
	WUT 6,	320,(V(K),K=189,194)	MATE159	NON-STANDARD
	WUT 6,	321,(V(K),K=195,19d)	MATE160	NON-STANJARD
	WUT 6,	1002	MATE161	NON-STANDARD
	WUT 6,	323	MATF1611	NUN-STANDARD
	WOT 6,	1002	MATE162	NON-STANDARD
340	WOT 6,	1040, NCUN,NTEMP,FLOW,NKUN,NMO,NDAY,NYEAR	MATE163	NUN-STANDARU
	WUT 6,	1050	MATE1641	NOG-STANDARD
	WUT 6,	1050	MATE1642	UPN-S LANDARD
	WUT 6,	1050	MATE1643	NUT-51A40A80
342	WUT 6,	384	MATE1644	NOM-5TANDARD
	WUT 6,	1055	VATE165	NOP-STATUARD
	WUT 6,	1002	VATE165	NOM-STANDARD
	WUT 6,	1002	MATE1661	ары-5Тийдак0
	WUT 6,	1005	MATE167	NULL-STANDARD
	WUT 6,	1002 •	MATE16/U	101-ST44UARD
361	FORMAT	(38X,2F5.2,2X,2F5.2,28X,3H 21)	4ATE1671	
362	FURMAT	(32X,2F5.2,2X,2F5.2,2X,2F5.2,22X,3U 20)	MATE1672	
363	FORMAT	(26X,2F5.2,2X,2F5.2,2H +2F5.2,2X,2F5.2,16X,3H 19)	MATE16/3	
364	FURMAT .	(20X,2F5.2,2X,2F5.2,2X,2F5.2,2X,2F5.2,2X,2F5.2,2Y,2F5.2,1UX,3H 18)	MAFE1674	
365	FURMAT	(19X, F5.2, 2X, 2F5.2, 2H +2F5.2, 2H +2F5.2, 2H +2F5.2, 2H +2F5.2,	MA151675	
1		9X, 3H 17)	MATE1676	
366	FORMAT	(20X,2F5.2,2H +2F5.2,2H +2F5.2,2H +2F5.2,2H +2F5.2,10X,	M4TE1677	
٦		3H 16)	MATE157+	
367	FORMAT	(14X,2F5.2,2H +2F5.2,2H +2F5.2,2X,2F5.2,2H +2F5.2,2H +	MATE1673	
-		2F5.2,4X,3H 15)	%ATE1679	
368	FURMAT	(13X,F5.2,2X,2F5.2,2H +2F5.2,2X,2F5.2,2X,2F5.2,2H +2F5.2,	MAFE1630	
T		2X,F5.2,3X,3H 14)	MATE1681	
369	FURMAT	<pre>(14X,2F5.2,2H +2F5.2,2X,2F5.2,2X,2F5.2,2X,2F5.2,2H +2F5.2</pre>	,MATE1682	

- 52 -

MATE,	A MANI	OLD TEST DATA REDUCTION PROGRAM. F.C. ENGEL, WAPD.	5/25/65	PAGE 6
1 370 1 371	FORMAT FURMAT	<pre>4x, 3H 13) (13x, F5.2, 2x, 2F5.2, 2H +2F5.2, 14x, 2F5.2, 2H +2F5.2, 2x, F5.2, 1 3x, 3H 12) 3x, 3H 12) (14x, 2F5.2, 2H +2F5.2, 2H •F5.2, 12x, F5.2, 2x, 2F5.2, 2H +2F5.2, 7H ** 11) 7H ** 11</pre>	MATE1663 MATE1684 MATE1684 MATE1665 MATE1686 MATE1687	
372 1 373	FURMAT FURMAT	<pre>(13x,F5.2,2x,2F5.2,2H +2F5.2,14X,2F5.2,2H +2F5.2,2X,F5.2, 1 3X,3H 10) (14x,2F5.2,2H +2F5.2,2X,2F5.2,2X,2F5.2,2X,2F5.2,2H +2F5.2,</pre>	MATE1688 MATE1689 MATE1690	
374 1	FURMAT	4x,3H U9) (13x,F5.2,2X,2F5.2,2H +2F5.2,2X,2F5.2,2X,2F5.2,2H +2F5.2, 0 2x.F5.2.3x.3H D9)	MATE1691 MATE1692 MATE1593	
375	FURMAT	(14X, 2F5.2, 2H +2F5.2,2H +2F5.2,2X,2F5.2,2H +2F5.2,2H + 2F5.2,4X,3H U7) 2F5.2,4X,3H U7)	MATE1694 MATE1695	
316 377	FURMAT	(ZUX,ZFD.Z,ZH +ZFD.Z,ZH +ZFD.Z,ZH +ZFD.Z,ZH +ZFD.Z,ZH +ZFD.Z,LUX) 3H 06) (19X,FD.Z,ZK,ZFD.Z,ZH +2FD.Z,ZH +2F5.Z,ZH +2F5.Z,ZY.F5.Z,	MATEl694 Matel694 Vatel697	
1 478	FURMAT	9X,3H 05) (20X.2E5.2.2X.2E5.2.2X.2E5.2.2.2E5.2.2.10X.3H 04)	4ATE1698 4111593	
379	FURMAT	(26X, 2F5, 2, 2X, 2F5, 2, 2H, +2F5, 2, 2X, 2F5, 2, 16X, 3H, 93) (26X, 2F5, 2, 2X, 2F5, 2, 2H, +2F5, 2, 2X, 2F5, 2, 16X, 3H, 93)	14TE1700	
181 181	FORMAT	132X,ZFD.Z,ZX,ZFD.Z,ZX,ZFD.Z,ZZX,3H UZ) (38X,2FD.Z,ZX,2F5.Z,23X,3H U1)	MATEL7J2	
ر8٤ 1		1824 QASTUV ARCUEFGALJAREMA QASTUV AX)	44151795 44TE1795	
334	FURMAT	(14X,42H * INLET ** JUTLET + TIE 30LT)	VATE1700 4.1.17.	
685	NUT 6,	361, (VAUKM(K),K=1,41 362, (VAUKM(K),K=5,10)	"ATEL/L VATEL72	N04-5 FN0A30
	WOT 6,	363, (VNORM(K),K=11,13)	42Tc173 44Tc175	V04-514×0×00
	WUT 6,	364, (VURAININALIA) 69 365, (VURA(N), K=29, 38)	4ATE175	404-51A00420
	WUT 6,	366, (VHURM(K),K=30,43) 347. [JANDAM(K),K=40,60)	MATE170 MATE177	404-51400480 404-5140480
	WUT 6,	363, (V:40AM(K), K=61, 72)	1214173	NUR-JI NOARD
	WOT 6,	369, (VNURM(K),K=73,64) 270, (VNU2M(K),K=86,94)	MATE179 Mate180	NUN-STANDARD
	WUT 6,	371, (VNORM(K), K=95, 104)	415101 1415101	05000412-100V
	WUT 0,	372, (VNORM(K), K=105,114)	44TE152 2+TE152	40.4-5.TAMDA.80
	WUT 6.	373, (VNUKM(K),K=127,120) 374, (VNURM(K),K=127,138)	4415105 44TE184	
	WUT 6,	375, (VNORM(K), K=139, 150)	4ATE185	NON-STARDARD
	WOT 6,	376, (VNURM(K),K=151,160)	MATE186	NON-STANDARD
	WUT 69	3//; (VNUKM(K);K=F5L;L/U) 378. (VNDRM(K).K=171.180)	4.41 E 1 4 7 4.4 T E 1 8 8	NUN-STANDARD
	WOT 6.	379, (VNDAM (K), K=131, 188)	4ATE189	NON-STANDARD

- 53 -

PAGE 7. NON-STANUARD NON-STANDARU NON-STANDARD NON-STANDARD NON-STANDARD NON-STANDARD NON-STANDARD VON-STANDARD NUN-STANDARD **VUN-STANDARD** NON-STANDARD 5/25/60 **MATE2003** 4ATE 2004 MATE2002 MATE2001 MATEZOS MATE206 MATE190 NATE192 **MATE197** MATE199 **AATE20**Ů MATE201 MATE207 MATE208 MATE209 MATE210 MATE211 MATE212 MATE214 MATE215 MATE216 MATE210 MATE219 44TE223 WATE 220 MATE213 4ATE220 4AT6224 44TE225 NATE233 4ATE235 MATE230 WATE191 4ATE221 4ATE223 42TE223 TATE230 4AT6232 NATE217 4ATE 22 44TE221 4ATE231 4ATE234 A MANIFOLD TEST DATA REDUCTION PROGRAM. F.C. ENGEL, WAPD. 3()||####B#####@######B###############])] 1040, NCON,NTEMP,FLOW,NRUN,NMD,NDAY,NYEAR 2913****5***5****5***5***5** 2814******************************** S 7 H * * * * 5 * * * 5 * * * 5 * * * 5 * * * 5 * * * 5 * * * 5 * * * 5 * * * 5 * * * 5 * * * * 5 * * * * 5 * * 26:1****5****5***5***5*** 254****5****5***5*** 2414****\$******************** 2314****\$**************** 2214****5****5****5** [3H***B*************** 18H××××5×××÷××××××××××× [7|++++5****5****5** 161****5***5***5*** 15:1************** (****\$***?****?*** 13H****5***5***) 380, [VNORM(K),K=189,194) 381, (VNORM(K), K=195,198) 124********** []||********** 10T******** 0×******* 8H****** (******HZ 6H****5*) 5H****\$) (****HV 3H***) 2H**) 1H*) (1H+28X 1H+28X 1H+28X 1H+24X [1H+28X 1H+28X 11+28X **1**H+28X (1H+28X (1H+28X (1H+28X (1H+28X 1H+28X (·1H+28X 1H+28X LH+28X 1H+28X 11+28X 1H+28X 1H+28X 1H+28X [1H+28X 1H+28X 1H+28X (1H+28X [1H+28X 1H+28X 1H+28X 1H+28X 1H+28X 1H+28X 1H+28X 1050 1002 6,1050 384 6,1050 6,1050 6, 1002 383 **6 6** 6, 6, **0** ç, FURMAT • • FURMAT FORMAT FURMAT FORMAT FURMAT FURMAT FURMAT FORMAT FURMAT FURMAT FORMAT FURMAT FURMAT FORMAT FURMAT FURMAT FORMAT FURMAT FURMAT FURMAT FURMAT WO1 WUT WUT MOT MUT MOT WOL WOT WO T NUT WUT MATE, 388 389 1115 1120 1125 1130 1135 140 1145 1150 1155 1160 1165 1170 1180 1185 190 1195 200 205 210 215 220 225 230 235 240 245 250 265 1110 255

PAGE 8

:

•

MATE,	A MANIFU	DLD TEST DATA REDUCTION PROSRAM. F.C. ENGEL, WAPD.	5/25/65
1270	FORMAT (1H+28X 33H***********************************	MATE237
1275	FURMAT 1	<u>14+23× 34H****5****5****5*****5*****5****5****</u>	MATE238
128J	FUXMAT (1.H+2.8K Э́́ЭНАААААБААААБААААБААААБААААБААААБАААА	MATE239
1285	FURYAT	(////36X3/H************************************	MATE240
1290	FURMAT (15X,F4.2,3H - ,F4.2)	MATE241
		00 410 N=1,24	MATE242
		(0 = (N))	MATE243
410	J	ONTINUE	MATE244
	Ľ	00 450 K=1,198	MATE245
	-	F (VNDRM(K)-0.75) 450,450,415	MATE246
415	1	F (1.230-VNORM(K))450,420,420	MATE247
420			MATE248
	U.	i0 TU (425,426,427,428,429,430,431,432,433,434,435,436,4	STMATE249
-1	•	438,439,440,441,442,443,444,445,445,446,447,448),NRVEL	MATE250
425	-	(11) =r (11) -r (10)	MATE251
	0	50 TO 450	MATE252
426		(C2)=L(V2)+L	MATE253
	ں ا	t0 10 450	MATE254
427		.(03)=L(03)+1	MATE255
		GU TU 450	MATE256
428	_	.{04)=L{u4}+l	MATE257
	0	50 ID 45u	MATE250
429		.(05)=L(05)+I	VATE259
	0	50 T0 450	MATE260
430		. (06) = L (06) + L	MATE261
	ر. ا	50 T0 450	MATE262
431		.(07)=L(07)+I	MATE263
		50 TU 450	MATE264
432		.(08)=L(09)+I	MATEZeS
	C	0 1 1 4 5 U	MATE265
433		(09)=L(09)+1	4ATE267
	J	50 10 450	HATEZ68
434		[+(C1)=F(C1)]	MATE269
		50 T0 450	MATEZTU
435	-	(11) = (11) + (11) + (11)	MATE271
	U.	50 T0 450	MATE272
436		.(12)=L(12)+1	4ATE273
	J	30 TO 450	MATE274
437	_	.(13)=L(13)+l	WATE275
	Ċ	60 T0 450	MATE276
438		.[14)=L(14)+1	MATE277
			MATEZ 78
439	<u>ب</u>	.(15)=L'(15)+1	MATE279

					•
MATE,	AM	ANI	FOLD TEST DATA REDUCTION PROSRAM. F.C. ENGEL, WAPD.	5/25/65	PAGE 9
			GO TO 450	MATE280	
440			L(16)=L(16)+1 GO TO 450	MATE282 MATE282	
441			L(17)=L(17)+1	MATE283	
			GU TO 450	MATE284	
442			L(18)=L(18)+1	MATEZ85 MATE284	
2.4.4			GU TU 450 1 (101=1 (101+1	MATE287	
0++			C1127-C11711 G0 T0 450	MATE288	
444			L(20)=L(20)+1	MATE289	
			GO TO 450	MATE290	
445			L(21)=L(21)+1	MALEZ91	
			GO TO 450	MATE292 MATE203	
446			L (22) =L (22) + L Certo 4 Eur	MATE294	
447			60 10 420 1 (23)=1 (23)+1	MATE295	
			C/ T/ 450	MATE296	
644			00 10 700 L (24) ≠L (24)+L	MATE291	
) - -			G0 T0 450	MATE298	
450			CONTINUE	MATE299	
	MUT	6,	1055	MATE300	NON-STANDARD
	WUT	6,	1050	MATE301	NUN-STANDARU
	MUT	6,	1050	MATE304	N NU-SIANDARD
	MUT	6,	1285	MATE305	NON-STANDARD
	WUT	61	1050	MATE306	NUN-STANDARD
	WOI	6,	1050	MATEBUT	NUN-STANDARD
			SU2Y = 0.75	MAIE303 Wate200	
			DO 499 N=1,24	MATE209	
			LN = L(N)	MATENI	
	TON	4	AREL = 5027 + 0.020 1200 SH7Y AMFT	MATE312	N NN-STANDARD
		0	IF 1 N 1 499.499.460	MATE313	
.460			G0 T0 [461, 462, 463, 464, 465, 466, 467, 468, 409, 470, 471, 472,	MATE314	
-			473,474,475,476,477,478,479,480,481,482,483,484,485,486,	MATE315	
2			487,488,489,490,491,452,493,494,495),LN	MATE316	
461	MOT	6,	1110	MATE31/	NUN-STANDARD
	i i		GU TU 499	MATES 10	N N-STANDARD
462	I DM	0	607 UI UD	MATE320	
463	MOT	6.	1120	MATE321	NUN-STANDARD
2			60 10 4,93	MATE322	
464	MUT	6,	1125	MATE323	NON-STANDA.40
			GU TO 499	MAIEJC4	

:

.

.

.

MATE,	A MA	ANIF	FULD TEST DATA REDUCTION PRUGRAM. F.C. ENGEL, WAPD.	5/25/65	PAGE 1
465	TOW:	44	1130 MAT	TE325 N	IDN STANDARD
466	NOT	• •	U 10 499 MAT CO TO YOO	TE327	ION - STANDARD
467	NUT	61.		16329 N	ION - STANDARD
468	MOT	6,	60 10 499 1145 MAT	1E331 N	ION -STANDARD
469	WOT	64	60 10 499 MAT GO TO 400 MAT	TE333	ION STANDARD
470	NOT	6,	MAT 1155 AAT CO TO 493	TE335	UN STANDARD
471	MUT	61	MAT 1160 MAT 2011 MAT 20111	TE337 N	IUN -STANDARD
472	MOT	6		TE339 N	ION - S T ANDARD
473	WOT	6,		TE341 P	ION -STANDARD
474	WŪT	6,	GU-1U 499 MAT 1175	TE343 N	IDN -STANDARD
475	NUT	6,	GO TO 499 MAT 1180 MAT	TE344 TE345	ICA STANDARD
476	MUT	• •	GO TO 499 MAT 1185 MAT	re346 1e347 N	ICN - STANDARD
477	WUT	• •	GU TO 499 MAT 1190	TE349 N	ION - STANDAKD
478	MOT	, ,	G0 T0 499 MAT 1195 MAT	re350 re351 ⁿ	ION-STANDARD
479	МОТ	6 1	GO TO 499 MAT 1200 - MAT	TE352 TE353 N	IGN-STANDARD
480	TÜW	6,	GO TO 499 MAT 1205 AAT 1205 AAT	18854 18855 18855	ION - STANDARD
481	WOT	6,	60 10 499 MAT 1210 MAT 261 TO 200 MAT	ГС 300 ГС 357 ГР 358	ION - STANDARD
482	MUT	6,	GU 10 725 1215 MAT GU TO 499 MAT	re359 N	ION - S TANDARD
483	WOT	6,	1220 60 TD 499 MAT	FE361 N FE362	ION-STANDARD
484	WOT	61	1225 MAT G0 T0 499 MAT	re363 N re364	IOI - STANDARD
485	WOT	69	MAT 20 20 TO 200	reads N	IOM-STANDARD
486	MOT	6,	1235 MAT	reada	ION-STANDARD

- 57 -

MAIC.	2	INVI	FULU IESI DAIA KEUULIIUN PRUSKAM. F.L. ENGEL, WAPU.	69/62/6	PAGE 11
			GO TO 499	MATE368	
487	HOT	\$9.	1240	MATE369	NDN-STANDARD
			GO TO 499	MATE370	
488	NUT	6		MATE371 Wate373	M JN-STANDARD
007	LOM	4	50 10 433 1250	MATE212 MATE272	
10 1		5	GO TO 499	MATE374	
490	MUT	9	1255	MATE375	MU-STANDARD
			GO TO 499	MATE376	
491	TOW	6	1260 Gn Tri 499	MATE377 MATF378	AJN-STANDARD
492	NUT	6,	1265	MATE379	N JN-STANDARD
÷			GO TO 499	MATE380	
493	WUT	6,	1270	MATE381	N JN-STANDARD
,			60 10 499	MATE382	
464	TOM	61	1275	NATE383	N N-STANDARD
	1		60 10 499	MA E 384	
402 402	1 DM	,		MA1E385 MATE385	NUN-STANDARD
5 C C L				MAIE330 Matrous	
000	1011	5		MATE201 Mate200	
5.01		0 4	1002 HOAN NTEAD FLOR ADIN AND NOAV NVEAD	MATE280	
1 C C C C C C C C C C C C C C C C C C C		0	LOHOF NOUNFIXTENTFELONFIXONFIXIOFADATFIXTEAN	MATEXON MATEXON	
0.0 2 1 1			10 001 N+1/170 16 (100-k) 202.601.600	MATEROCI	
400			LT (100-K) 00540014000 GD TD(610.610.610.610.610.610.620.620.510.610.610.	MATEROL	
			620.520.620.620.610.610.610.610.620.620.630.630.620.620.	MATE 542	
• ~			610.510.010.620.620.630.630.630.530.520.620.610.620.620.	MATE393	
نہ ہ			630,630,640,640,630,630,620,620,620,610,620,630,630,640,640.	MATE394	
t -			649.640.630.630.620.610.610.620.620.630.640.640.650.650.640.	MATE395	
· in			640.533.620.610.610.620.630.040.650.650.650.650.643.630.	MATE396	
. .)			620, 610, 610, 620, 630, 640, 650, 650, 640, 630, 620, 610, 610, 620,	MATE397	
7			630,640,050),K	4A TE 398	
601			GO TO 650	MATE3981	
602			M = (K - 100)	MATE3982	
603			GO TO. (640,630,620,610,610,620,630,640,650,650,	MAIE3983 MAIE3983	
			640, 630, 620, 610, 610, 620, 630; 640, 650, 650, 650, 650, 640, 630,	MATE399	
2			620,610,610,620,630,640,640,650,650,640,640,630,620,610,	MA TE 400	
'n			610,620,630,630,640,640,640,640,640,630,630,620,610,620,620,	MATE401	
4			630,630,640,640,630,620,620,620,610,620,620,630,630,630,630,	MATE402	
Л			630,620,620,610,610,615,620,620,630,630,520,520,510,510,	MATE403	
0			610,610,620,620,620,620,610,610,610,610,620,620,610,610,	MATE404	
1			610,610,610,610),M	MATE405	
61 0			A≖1.2816	MATE406	

•

- 58 -

\sim	
ല	

.

MATE,	A M.	NNI	OLO TEST DATA REDUCTION PROGRAM. F.C. ENGEL, WAPD.	5/25/65	PAGE 12
			GU TO 800	MATE407	
620			A=1.0477	MATE408	
			GO TO 300	MATE409	
630			A=0.9049	MATE410	
			GO TO 800	MATE411	
640			A=0.8382	MATE412	
			GO TO 800	MATE413	
650			A=0.8039	MATE414	
			GO TO 800	MATE415	
800			CONTINUE	MATE416	
801			VW(K)=VNORM(K)*A	MATE417	
803	TOW	6	1055	MATE418 NCN	I-STANDARD
804	MUT	, 9	1002	MATE419 NON	I STANDARD
805	10%	6,	1100	MATE420 NON	ISTANDARD
1100	FOR	MAT	(40H0 EFFECTIVE NURMALIZED FLOW DISTRIBUTION)	MATE421	
302			CUNTINUE	MATE422	
	WUT	, 0	1002	MATE4221 NON	I STANUARD
860	WUT	9	361,(VW(K),K=1,4)	MATE423 NDM	I STANDARD
	MOT	0	362,(VW(K),K=5,10)	MATE424 NON	STAWDARD
	MUT	6,	363, (VW(K), K=11,18)	MATE425 NON	I-STANDARD
	MUT	, 0	364f(VW(K),K=19,28)	MATE426 VON	I STANDARD
	NUT	6.	365, [VW(K), K=29, 38)	MATE427 NON	ISTANDARD
	MUT	0	366, (VW(K),K=39,48)	MATE428 NON	STANDARU
	WUT	6,	367, [Vw(K),K=49,60]	MATE429 NON	F STANDARD
	TOW	6,	368, (Vw(K),K=61,72)	4ATE430 NON	1-5TANJA30
	WUT	6,	369, [VW(K),K=73,84)	MATE431 NON	I-STANDARD
	WOT	• •	370, [VW(K),K=85,94)	MATE432 VOV	I-STANCARU
	NOT	• 9	371, [VW(K),K=95,104]	MATE433 NUV	USEUNARD
	NUT	\$	372,[VW(K),K=105,114)	MATE434 NOH	UNECNATO -
	WOT	6,	373, (VW(K),K=115,126)	MATE435 MON	- 5TAW0AX0
	NUT	6,	374, (VW(K),K=127,138)	MATE436 NOV	- STANDARU
	NUT	6,	375,(VW(K),K=139,150)	MATE437 NUN	- STANUARD
	WOT	• •	376, (VW(K),K=151,160)	MATE433 NUN	-STANJAKD
	WUT	•	377,(VW(K),K=161,170)	MATE439 NON-	 STANDARD
	WOI	6,	378, (VW(K),K=171,180)	MATE440 NUV-	– S F ANJARD
	WOT	• •	379, (VW(K),K=181,1d8)	MATE441 NON	.5TANDARD
	WOT	\$	380, (VW(K),K=189,194)	MATE44 NO.4	5 T ANC 4.7.0
	NOT	\$	381,(VW(K),K=195,198)	MATE443 NUN	S TANJARD
	WOT	6 ,	1050	MATE444 NON-	· STANJARD
861	MUT	•	333	MATE4441 40N-	S T AN DARD
	WOI	6,	1002	MATE445 NON-	-STANDARU
886	WOT	6,	1040, NCUN,NTEMP,FLOW,NRUN,NMU,NDAY,NYEAR	MATE446 NON-	- STANDARD
	WUT	6,	1002	MATE447 NUN-	STANÜARD

13	
PAGE	40N-STANDARD 40N-STANDARD
5/25/65	MATE4473 MATE4481 MATE448 MATE449
WAPD.	
ENGEL .	
F.C.	
REDUCTION PROGRAM.	
DATA	
A MANIFOLD TEST	WUT 6,1050 WOT 6, 384 CONTINUE GG TO 10 END
ATE,	888 889 900 900

.

.

•

...

•

NNHHHHHHHHHOCOCOCOCO	
• • • • • • • • • • • • • • • • • • •	
	× -
	.
6 9 6 9 8 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9	>
	Þ
910010806	
	S
34 513000	¢
	σ
	٩
	0
	÷
	Σ
40 40 4 40 4 4 4 4 4 4 4 4 4 4 4 4 4 4	
	¥
	_
	I
	9
	L
6 0 0 0 4 6 9 0 	٥
	J
6 8 8	
8 8 0 8 0 8 0 8 0 8 0 8 0 8 0 8 0 8 0 8 0 8 0 8 0 8 0 8 0 8 0 0 1 1 1 1 1 1 1 1 1 1 1 1 1	8
	4

-0000r054m0-050r054m0-

MANIFOLD DATA REDUCTION, F.C.ENGEL, APD

TUBE VELOCITY DISTRIBUTION, NORMALIZED

Figure 3

- 61 -

÷ # # INLET

TIE BOLT OUTLET

65

6 15

DATE

10

11

NRUN

FLOW = 305.0

78

Ħ

NTEMP

-

Ħ

NCON

MANIFOLD DATA REDUCTION, F.C.ENGEL, APD

٠.

******* FLOW HISTOGRAM *******

				5****	****	\$** ***	********	\$***	1)************************************						
* * *	* * * *	* *	* *	* *		* * *	* * * * *	* * *	* * * * *	*	*		*		
0.77 * 0.79 *** 0.81 0.83	0.85 ***	0.89 *	0.93 ##	0.95 **	**** 65*0	1.01 ****	1.03 **** 1.05 ****	1.07 ****	1.09 **** 1.11 ****	1.13 *	1.15 *	1.17	I.19 **	1.21	I•23

15 65

Ŷ

DATE

10

NRUN =

FLOW = 305.0

78

NTEMP =

NCON =

Figure 4 - 62 -

× 0 1.0 0. . د ت . ن £ **C** . 98 0.97 > • 0 • උ ⊃ а; 6 • 05 • 02 .01 • 02 0.98 •09 0.98 S C 0.84 1.04 - C -.01 0.96 α 0 • 0 • 0.0 • Ċ 0 0 C 0.99 [0•T 0.38 L • 03 . **a**. 0.95 • . . . • 0 С. . ت C C ..57 4 ပ် **ن** 2. S \odot **ن** C + 5. .01 .01 02 0.92 _ Ō ¥ •0 • ి • ...CO C. 59 300 0.95 7 C.98 1.01 C.98 σ C • 98 0.99 5 99 C. 8C I G 1.00 0.96 щ 6.0 0 :0 • . \circ w 5° 5° 10° C.8C C.58 C.92 ပ် ני י) C.98 1.09 1.11 പ S 0.87 1.03 C.84 1.00 0. C.98 æ . С 4

σ ω Ŷ 5 4 12 11 10 09 08 ξ

MANIFOLD CATA REDUCTION, F.C.ENGEL, APD

TUBE VELOCITY CISTRIBUTION, NORMALIZED

Figure 5 - 63 -

----11 NCON

65

~

CATE

12

0

NRLN

602.C

0

FLON

с С

11

NTEMP

CUTLET * * INLET

RCLT

u L L

*

MANIFCLD CATA REDUCTION, F.C.ENGEL, APD

.

********* FLOW HISTOGRAM ********

		**		**	**		**	**	*	****	************	**********	****	****	•	****	G****				**	*		
	.77 - 0.79	.79 - 0.81	-81 - 0.83	3.83 - 0.85	1.850.87	.87 - 0.89	0.89 - 0.51	:•91 - 0.53	0.93 - 0.55		65.0 - 76.	.9-9 - 1.Cl	•01 - 1.C3	.03 - 1.05	.05 - 1.07	•07 = 1.69	03 - 1.11	.11 - 1.13	.13 - 1.15	.15 - 1.17	17 - 1.19	.19 - 1.21	21 - 1.23	
с	0	O	S	0	Ö	o	0	J		υ	C	J						1					Ч	

ç

~

~

DATE

12

н

ARUN

FLCh = 6C2.0

S С

H

NTEMP

NCON =

Figure 6

- 64 -
| * | | |
|--|-----------------|---------|
| | ^
3 | |
| $\begin{array}{cccccccccccccccccccccccccccccccccccc$ | °
∧ | 65 |
| $\begin{array}{cccccccccccccccccccccccccccccccccccc$ | S T | 7 7 |
| | 3 | DATE |
| 1 • 0 • 4
0 • 9 • 0
1 • 0 • 9 • 0
0 • 8 • 0
0 • 9 • 0
0 • 9 • 0
0 • 9 • 0
0 • 9 • 0
0 • 1 • 0
0 • 1 • 0
0 • 0 • 0
0 | d .
D | 13 |
| - C C C C C C C C C C C C C C C C C C C | 2 | =
VN |
| π π </td <td>ר
א</td> <td>Ž</td> | ר
א | Ž |
| | ר <u>ו</u> | 70.0 |
| | н
9 | = M01; |
| | u.
w | C |
| | ے
د | |
| 0 | A B | NTEMP |
| | | - |
| | | NCON = |

1004550100100455018001 *

MANIFOLD CATA REDUCTION, F.C.ENGEL, APD

TUBE VELOCITY CISTRIBUTION, NORMALIZED

Figure 7 - 65 -

TΙE + CUTLET *

BCLT

INLET

.

MANIFOLD CATA REDUCTION, F.C.ENGEL, APD

******* FLCW HISTOGRAM *******

	*	**		***		***	*	* * * *	* *	****	****	****	************		**	*********	******	****		¥		*		
0.77	0.79	0.81	0.83	0.85	0.87	0.89	1.5.0	0.53	0.55	L-5 • 0	6500	1.C1	1.03	1.C5	1.07	1.09	1.1.1	1.13	1.15	1.17	1.19	1.21	l.23	
I	ł	ł	ł	I	ł	Ŧ	I	ł.	1	I	1	I	ł	ł	ŧ	ł	ł	ł	t	ł	I	1	I	
0.75	0.77	c.79	C.81	C.83	C.85	C.87	C.89.	16.0	C.93	C.95	C.97	C• 5 • 0	1.01	1.03	1.05	I. 07	1.09	1.11	1.13	1.15	1.17	1.19	1.21	

έS

CATE

13

H

NRUN

70.C

**

FLOW

80

11

NTEMP

-

NCON =

- 66 -

20 9829548 ο 60 08 07 06 **1.82** 1.01 0.62 66°C •00 × • <u>،</u> • . 3 .07 0.96 0.95 0.88 • 0 • ⊃ 1.05 08 •06 • • ò 0.83 .08 • 05 ¢ • • σ .12 0.97 ٩ • • 0 • • 05 •08 0.99 0.75 Z ō x 5. - 96 0.95 _ C • . • • 0 • ¥ - 98 0.98 o • ō 1.01 1.00 0.96 ō 0 Ŧ • • c 1.10 0.99 L 8 • • w 1.02 0. 0.86 1.02 •0• • 0 5 0.94 1.06 •03 0.96 0 • • • . ပ 0.98 0. 1.08 0.85 0.99 0.94 0.94 80 •

MANIFOLD DATA REDUCTION, F.C.ENGEL,APD

TUBE VELOCITY DISTRIBUTION, NORMALIZED

TIE BOLT OUTLET * **INLET**

.

6.9

15

DATE

15

Ħ

NRUN

310.0

H

FLOW

80

Ħ

NTEMP

-

NCON

Figure 9

- 67 -

MANIFOLD DATA REDUCTION, F.C.ENGEL, APD

******* FLOW HISTOGRAM ********

•	* *	÷.	* * *	÷	****	******	的中年年来的年年年年	*******	*****	******	*****	********	* *	* =		ŧ			
0.77 0.79 0.81	0.83 0.85	0.87	0.89 0.91	0.93	56° 0.	16.0	66.0	1.01	1.03	1.05	1.07	1.09	1.11	1.13	1.15	1.17	1.19	1.21	1.23
	1 1	I	1 1	I	1	ŧ	ŧ	t	ł	ì	1	ŧ	1	ł	1	1		1	ł
0.75 0.77 0.79	0.81 0.83	0.85	0.89	0.91	0.93	0.95	0.97	0.99	1.01	1.03	1.05	1.07	1.09	1.11	1.13	1.15	1.17	1.19	1.21

7 15 65

DATE

15

Ħ

NNUN

FLOW = 310.0

80

NTEMP =

NCON =

Figure 10

- 68 -

APPENDIX B

CAVITATION IN INLET ELBOWS

During the preliminary flow distribution tests the flow cavitated in the inlet and outlet short radius elbows. This manifested itself by a surging noise in the flow of about one cycle per second frequency and considerable vibration of the model. At 300 gpm the cavitation could be nearly suppressed by raising the back (outlet) pressure to 15 psi. At 450 gpm it continued full force with 10 psi back pressure. Observations with the help of a stethoscope pinpointed the source of the noise to be located at the centers (45°) of the two 90° inlet PVC elbows. These plastic (PVC) elbows were installed to give greater resiliancy and thus prevent cracking of the glued joint at the model inlet. The inlet elbows were replaced by long radius schedule 10 stainless steel elbows, which had been proposed earlier for the reference design. The outlet elbows were replaced by a straight pipe connection.

It had been assumed that the PVC elbows were of standard short radius schedule 40 dimensions. However it turned out on disassembly that their inside bend radius was practically zero, i.e. a sharp corner.

Assuming a radius of 1/32 inch and a flow profile approaching that of a free Vortex/5 ($V_{\Delta P}$ = constant) it could be calculated for the case of the elbow with a shrap inside turning radius, using the expressions and curve of TMB 575/6, that water, fairly well saturated with air because of the free interface in the supply tank, would have a static pressure well below the vapor pressure. Thus steady state cavitations should be expected. For the long radius elbows the static pressure was estimated by the same method to be 15 psi above vapor pressure so that there should be no cavitation. For the standard schedule 40 short radius elbow the static pressure was estimated to be 10 psi above vapor pressure. Here again no cavitation would be expected.

The same results were obtained by the use of the equation of the velocity profile in a bend contained in the paper of W. A. Marris, /7

$$p = \frac{Vm^2}{\log_e \frac{p}{R_i}} \times \frac{d^2}{\frac{1}{R_i^2}} = \frac{1}{\frac{1}{R_i^2}} + \frac{1}{\frac{1}{R_i^2}} = \frac{1}{\frac{1}{R_i^2}}$$

- 69 -

where:

p = static pressure at location x

R_i = inside radius of bend R_o = outside radius of bend

 $d = width of channel = R_0 - R_1$

x = radial distance of location from the inside boundary

$$\mathbf{r} = \mathbf{R} + \mathbf{x} = \mathbf{radial}$$
 distance to the location under consideration

APPENDIX C

DERIVATION AND USE OF MIXING LENGTH

from F. Sjenitzer - The Pipe Line engineer 1958)⁴ K = effective axial diffusivity, m² sec⁻¹ a = pipe radius, m V = $\frac{L}{t}$ = mean velocity in pipe, m sec⁻¹ L = pipe length, m t = time for passage through pipe, sec s = length of pipe containing contaminated liquid, m y = fractional purity at either end of mixing zone v = kinematic viscosity of SO/SO mixture, m² sec⁻¹ γ = fanning friction factor

G. I. Taylor found that the ratio $\frac{K}{aV}$ depend solely on the fanning friction factor

$$\frac{K}{aV} = 10.1 \sqrt{\frac{1}{2}\gamma} - 7.15 \sqrt{\frac{1}{2}}$$

then

$$\frac{K}{aV} = c \gamma^{n} \left(\frac{L}{a}\right)^{m}$$
where $c = 2.4 \times 10^{7}$
 $n = 3.6$
 $m = 0.141$

- 71 -

Combining and rearranging gives

$$S = 1.96 \times 10^{4} (\frac{L}{a})^{0.07} \sqrt{aL} - \gamma^{1.8} erf (2y-1)$$

For inner poison tube

L = 4.5 ft a = $\frac{0.340}{2}$ in. U = 10.7 ft/sec $\gamma = 0.0060$ erf (2y-1) = erf (0.99) = 1.640 Total mixing length S = 1.30 ft Forward diffusion = $\frac{S}{2}$ = 0.65 ft 0.65 ft/10.7 ft/sec = 0.061 sec forward diffusion time To obtain poison arrival at the top of the tubes find from test data:

- (1) time after injection for first arrival of poison at bottom
- (2) time for completion of concentration change at bottom
- (3) the mean of 1 and 2, i.e. the time for the front to arrive at bottom of tube.
- (4) transient time through tube; at 600 gpm 4.5 ft ÷ 10.7 ft + sec
 = 0.42 sec.

(5) forward diffusion length at top of tube

= forward diffusion length at bottom minus change of diffusion length of in tube

$$\frac{s}{2} = \frac{(1) - (2)}{2} - 0.61$$
 seconds at 10.7 ft/sec

(6) time for arrival at the top of the tubes equals
 arrival of the front at the bottom of the tube minus
 transit time through tube minus forward diffusion length.
 For 600 gpm this is:

$$\frac{(1) + (2)}{2} - 0.42 - \left(\frac{(1) - (2)}{2} - 0.61\right)$$

APPENDIX D

REFERENCES

- 1. NASA CR-54291 (WCAP-2690), "Analysis of Reference System, Task I, Feasibility of a Chemical Poison Loop System, February 1965."
- 2. NASA CR-54420 (WCAP-2803), "Design of Laboratory Tests, Task II, Feasibility of a Chemical Poison Loop System, June 1965."
- 3. A. Acrivos, et al., "Flow Distribution in Manifolds," Chem. Eng. Sc. 10, 1959, p. 112.
- 4. F. Sjenitzer, "How Much do Products Mix in a Pipe Line?", The Pipeline Engineer, December 1958, D31, p. 615.11.
- 5. Hunsaker & Rightmire, Engineering Application of Fluid Mechanics page 151, McGraw Hill 1947.
- 6. S. F. Crump, "Determination of Critical Pressures for the Inception of Cavitation in Fresh and Salt Water as Influenced by the Air Content of Water," Figure 13 TMB 575, October 1947, NS 713 - 065.
- 7. A. W. Marris, "Radial Distribution of Temporal-Mean Peripheral Velocity and Pressure for Fully Developed Turbulent Flow in Curved Channels," p. 528, TR. A.S.M.E., Jrl. Basic Engrg. September 1960.

Distribution List

NASA Lewis Research Center (3 & repro) 21000 Brookpark Road Cleveland, Ohio 44135 Attention: Morton H. Krasner MS 49-2

NASA Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attention: Technical Utilization Office MS 3-16

NASA Lewis Research Center (2) 21000 Brookpark Road Cleveland, Ohio 44135 Attention: Library

U. S. Atomic Energy Commission (3) Technical Reports Library Washington, D. C.

Space Nuclear Prop. Office (2) U. S. Atomic Energy Commission Washington, D. C. 20545

Westinghouse Electric Corp. (1) Astronuclear Laboratory P. O. Box 10864 Pittsburgh 36, Pennsylvania

North American Aviation, Inc. (1) Rocketdyne Division 6633 Canoga Avenue Canoga Park, California

North American Aviation, Inc. (1) Atomics International Division 8900 Desota Avenue Canoga Park, California

General Atomic Div. (1) General Dymanics Corp. P. O. Box 1111 San Diego, California 92112 Attention: Chief, Tech. Inf. Services NASA Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attention: Thomas J. Flanagan MS 54-1 Contracting Officer

NASA Scientific and Technical Information Facility (6 and repro) P. 0. Box 5700 Bethesda, Maryland Attention: NASA Representative

NASA Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attention: Report Control Office

U. S. Atomic Energy Commission (3) Technical Information Service Extension P. O. Box 62 Oak Ridge, Tennessee

NASA Lewis Research Center 21000 Brookpark Road Cleveland, Ohio 44135 Attention: Nuclear Rocket Technology Office MS 54-1 Mr. Leroy V. Humble MS 49-2 (1) Mr. Samuel J. Kaufman MS 49-2 (1) Mr. Edward Lantz MS 49-2 (1) Mr. Harry W. Davison MS 49-2 (1) Mr. Richard L. Puthoff MS 49-2 (1) Mr. Herbert J. Heopler MS 54-1 (1) Dr. John C. Liwosz MS 54-1 (1)

American Machine & Foundry Co. (1) Government Products Group Greenwich/Stamford Complex 737 Canal Street Stamford, Connecticut

Distribution List (Continued)

NASA Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: Office of Reliability and Quality Assurance

NASA Ames Research Center (1) Moffett Field, California 94035 Attn: Library

•

NASA Flight Research Center (1) P. O. Box 273 Edwards, California 93523 Attn: Librarý

NASA Goddard Space Flight Center (1) Greenbelt, Maryland 20771 Attn: Library

Jet Propulsion Laboratory (1) 4800 Oak Grove Drive Pasadena, California 91103 Attn: Library

NASA Langley Research Center (1) Langley Station Hampton, Virginia 23365 Attn: Library

NASA Manned Spacecraft Center (1) Houston, Texas 77001 Attn: Library

NASA Marshall Space Flight Center (1) Huntsville, Alabama 35812 Attn: Library

NASA Western Operations (1) 150 Pico Boulevard Santa Monica, California 90406 Attn: Library