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# GOODYEAR AEROSPACE CORPORATION

AKRON 15, OHIO

# AN INDICATED SPECULAR DEGRADATION

# RATE FOR ALUMINIZED MYLAR SURFACES IN

# NEAR-EARTH ORBIT FROM RECENT PHOTOMETRIC

# OBSERVATIONS OF THE ECHO I SATELLITE

by Richard H. Emmons Goodyear Aerospace Corporation

(A paper presented before the 116th meeting of the American Astronomical Society in Flagstaff, Ariz., 24-27 June 1964.)

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Photometric measurements were made of the Echo I satellite with a visual comparison photometer on 1 March 1964 (Universal Time).

The data, taken over a wide range of phase angles (143 to 32 deg) and normalized for slant range, are best fitted in illuminance versus Russell's phase function by a surface model that is 96 percent specular and 4 percent diffuse. This indicates that the actual specular degradation rate for aluminized Mylar surfaces in a near-earth orbit, due to the combined effects of all environmental factors, is less than 1 percent per year.

If a reflectivity coefficient of 0.83 is assumed, the Echo I photometric observations yield a mean radius of curvature of 47 ft with a probable error of 3 ft. Several surface anomalies were observed having local radii of curvature ranging from 39 to 66 ft.

Together with the results from laboratory hypervelocity impact tests of very small particles on aluminized Mylar, the present indications may prove of value in further defining the micrometeoroid environment at Echo I's orbital height, either with respect to the flux in the vicinity of the mass cutoff or the mass cutoff itself. The present indications should also prove of value in predicting the useful life in nearby space of exposed optical and thermal balance surfaces.

The photometric studies of Echo I are being extended at Goodyear Aerospace Corporation under NASA-Langley Research Center Contract NAS 1-3114, monitored by William J. O'Sullivan. While final results from these additional visual and photoelectric measurements are not available at this time, the preliminary results appear to support the conclusion herein that Echo I's surface remains highly specular.

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Photometric measurements were made of the Echo I satellite with a visual comparison photometer on 1 March 1964 (Universal Time).

The data, taken over a wide range of phase angles (143 to 32 deg) and normalized for slant range, are best fitted in illuminance versus Russell's phase function by a surface model that is 96 percent specular and 4 percent diffuse. This indicates that the actual specular degradation rate for aluminized Mylar surfaces in a near-earth orbit, due to the combined effects of all environmental factors, is less than 1 percent per year.

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AUTHOR

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## SECTION I - INTRODUCTION

To obtain photometric data for bright artificial satellites, a hand-held, polarizing, visual-comparison photometer was designed and constructed at Goodyear Aerospace Corporation in January, 1964<sup>a</sup> (see Figure 1). The digital readouts (polarizing filter rotations) are converted to stellar magnitudes by means of the calibration determined from repeated observations of stars of known visual magnitude. <sup>1, b</sup> Allowances are made for atmospheric extinction according to the instantaneous zenith distances (z) of the reference stars and target.

With this instrument 21 photometric measurements were made of Echo I from North Canton, Ohio, (Smithsonian Astrophysical Observatory satellite tracking station No. 8544) in a 13.6-min period, during a high pass of this satellite on 1 March 1964 Universal Time (U. T.). Figure 2 was drawn from a oneminute exposure photograph taken at 0 hr 37 min U. T., 1 March 1964, showing Echo I's path left of the planets Venus and Jupiter as the first photometric observations were made. Also shown are selected star and constellation locations, right ascension/declination reference grid, and time tics.

Calibration observations of stars were made before and after the satellite date was taken, yielding a total of 39 measurements at 18 different elevations of 17 different reference magnitudes throughout the photometer's range. A least squares reduction was made of this data, providing a mean extinction coefficient of 0.207 and the instrument's calibration equation:

<sup>&</sup>lt;sup>a</sup>This photometer was first tried in its automatic record mode to measure Echo II's optical characteristics during its first week in orbit. The measurements obtained were consistent with predicted values that would result from a successful deployment of this satellite.

<sup>&</sup>lt;sup>b</sup>Superior numbers in the text refer to items in the List of References.

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Figure 1 - Visual Comparison Photometer for Bright Satellites

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Figure 2 - Echo I's Apparent Path Past Jupiter and Venus

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 $m_{2} + 0.207X = 6.446 - 3.802 \log (instrument reading)$ ,

where

 $m_0$  = the extra atmosphere magnitude, and

X = number of atmospheres ( $\cong$  sec z).<sup>a</sup>

The standard deviation of the calibration data was 0.196 stellar magnitude, which includes the effect of possible extinction changes. The correlation coefficient was 0.9836.

Table I gives the Universal Times (obtained via short-wave radio from CHU and tape recorded together with the photometer readings), the instrument readings, the phase angle, the reduced stellar magnitudes normalized to a 1000-stat-mi zenith range, and the indicated radius of curvature. The stellar magnitudes have been corrected for earthshine.<sup>2,3</sup> At the tracking station where these data were taken, the longitude was 81 deg 24 min 42 sec west, the latitude was 40 deg 52 min 45 sec north, and the elevation was 350 m above sea level.

The phase angles and geometrical circumstances of these observations were reduced from orbital elements provided by the Smithsonian Astrophysical Observatory, which were found to conform to observed times and positions obtained during the pass. The solar coordinates at this time were: right ascension, 22 hr, 48 min; declination, -7 deg 39 min.

The normalized stellar magnitudes given in Table I are plotted versus phase angle in Figure 3. Upon inspection of the data it was judged that the three unnumbered photometric values were the result of surface macrotexture anomalies, and they were therefore excluded from the microtexture analysis. The remaining 18 numbered data points, used in the subsequent microtexture analysis, yield a mean normalized stellar magnitude of +0.22.

<sup>a</sup>Actually used X = sec z - 0.0018167(sec z - 1) - 0.002875(sec z - 1)<sup>2</sup> - 0.0008083(sec z - 1)<sup>3</sup> (see Reference 4).

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TABLE ]	

ted for	ni and correc	00 stat r	ange of 10	to a slant r	e normalized	tmosphere magnitude	* Extra a
51.2	-0.048	31.7	1779	19.57	16.5	0 hr 49 min 49 sec	18
48.4	0.073	32.3	1693	21.92	16.95	0 hr 49 min 22 sec	17
38.7	0.503	35.4	1458	29.68	17.3	0 hr 48 min 05 sec	16
48.7	0.060	40.2	1276	37.77	28.3	0 hr 47 min 00 sec	15
45.2	0.211	43.6	1194	42.40	28.7	0 hr 46 min 28 sec	14
47.9	0.105	50.2	1081	50.57	35.7	0 hr 45 min 38 sec	13
49.6	0.040	54.2	1033	55.04	39.8	0 hr 45 min 13 sec	12
66.2	-0.553 <sup>‡</sup>	59.3	. 987	60.22	61.0	0 hr 44 min 45 sec	• •
57.1	-0.242 <sup>‡</sup>	64.5	953	64.97	53.2	0 hr 44 min 19 sec	
40.8	0.457	71.7	924	70.18	36.5	0 hr 43 min 46 sec	11
43.2	0.349	77.9	912	72.65	39.7	0 hr 43 min 19 sec	10
43.5	0.344	85.0	913	72.26	39.8	0 hr 42 min 49 sec	6
57.8	-0.247*	92.2	928	68.65	55.5	0 hr 42 min 18 sec	
44.4	0.324	105.9	1002	57.31	35.0	0 hr 41 min 15 sec	8
43.5	0.378	114.3	1085	49.07	30.0	0 hr 40 min 30 sec	7
46.3	0.247	118.6	1142	44.67	30.0	0 hr 40 min 04 sec	9
44.8	0.320	121.3	1184	41.80	27.1	0 hr 39 min 46 sec	5
48.1	0.171	127.3	1298	35.43	25.7	0 hr 39 min 02 sec	4
50.9	0.052	134.5	1486	27.37	22.0	0 hr 37 min 56 sec	ŝ
49.7	0.106	138.0	1606	23.32	18.5	0 hr 37 min 17 sec	7
45.8	0.280	142.8	1810	17.54	13.0	0 hr 36 min 13 sec	•
Radius of curvature (ft) <sup>+</sup>	Normalized stellar magnitude <sup>*</sup>	Phase angle (deg)	Slant range (stat mi)	Elevation (deg)	Instrument reading	Universal time	Number
		i	č				

1 ange Extra atmosphere earthshine.

<sup>+</sup>Assumes a coefficient of reflectivity of 0.83.

 $^{\ddagger}$ Excessive brightness attributed to a gross surface anomaly (macrotexture) and therefore is omitted from the microtexture analysis.

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Figure 3 - Echo I normalized Magnitudes versus Phase Angle

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### SECTION II - SURFACE MICROTEXTURE ANALYSIS

A regression analysis was performed on the 18 normalized magnitudes (m) with a IBM 1410 computer. The dependence of their corresponding illuminances (E) on the Russell<sup>5</sup> phase function  $[F(\psi)]$ , was investigated for a diffusely reflecting sphere that obeys Lambert's law. This analysis was based on the following regression equation:

$$\frac{E}{E_0} = \text{antilog}(-0.4 \text{ m})$$
$$= \frac{A}{4} + \frac{2}{3} \text{ BF}(\psi) ,$$

where

 $E/E_{o}$  = illuminance ratio,

A = weighting factor for optically specular reflection,

B = weighting factor for optically diffuse reflection,

$$F(\psi) = \frac{1}{\pi} \left[ \sin \psi + (\pi - \psi) \cos \psi \right], \text{ and}$$

 $\psi$  = phase angle (0 deg when satellite is opposite sun).

A least squares best-fit solution of this regression equation for the normalized data yields:

$$A = 3.13621,$$
  
 $B = 0.14424.$ 

correlation coefficient = 0.27,

and

standard deviation,  $\sigma = \pm 0.15$ .

This solution, transformed to stellar magnitudes, is plotted in Figure 3, together with curves for best-fit solutions where A = 0 (surface 100 percent diffuse), and B = 0 (surface 100 percent specular).

The present optical specularity  $(P_s)$ , not to be confused with the coefficient of reflection, is indicated by:

$$P_s = \frac{A}{A+B} = 96 \text{ percent}.$$

If the presently indicated optical specularity (96 percent) of Echo I is compared with its initial specularity at the time of deployment in orbit (12 August 1960), an actual rate of specular degradation of aluminized Mylar due to the simultaneously combined effects of all environmental factors in a near-earth orbit is inferred. The initial optical specularity ( $P_s$ ) of Echo I was close to 98 percent. On this basis the actual rate of specular degradation for aluminized Mylar in a near-earth orbit is indicated to be less than 1 percent per year:

1 - antilog 
$$\left(\frac{\log P_s - \log P_s}{3.55 \text{ years}}\right) \equiv 0.6 \text{ percent per year}$$
.

The photometric studies of Echo I are being extended at Goodyear Aerospace Corporation under NASA-Langley Research Center Contract NAS 1-3114, monitored by William J. O'Sullivan. While final results from these additional visual and photoelectric measurements are not available at this time, the preliminary results appear to support the conclusion herein that Echo I's surface remains highly specular.

#### SECTION III - SURFACE MACROTEXTURE ANALYSIS

The photometric data were also used to determine mean and local radii of curvature of Echo I's surface. The radius of curvature of a spherical specular surface having a coefficient of reflectivity of 0.83 (typical for aluminized My-lar) can be found from the relation:  $^{6}$ 

$$R_{c} = antilog(1.71806 - 0.2m_{sp})$$
,

where

 $R_{a}$  = radius of curvature in feet, and

To apply the above relation in the macrotexture analysis, it was first necessary to eliminate the contribution of diffuse reflectivity,  $2/3 \text{ BF}(\psi)$ , from the normalized magnitudes. From the previous regression analysis it can be seen that the representative specular normalized magnitude is:

$$\frac{\log \frac{A}{4}}{-0.4} = +0.264 ,$$

which yields a mean  $R_{0}$  of 46.3 ft.

The normalized magnitudes at phase angles of 92.2, 59.3, and 35.4 deg were selected for local radii of curvature determinations because they corresponded to independent data extremes, each at least 2  $\sigma$  from the best-fit and adopted regression curve. These magnitudes were reduced in illuminance by the amount  $2/3 \text{ BF}(\psi)$  to yield m<sub>sp</sub> = -0.22, -0.51, and +0.65, respectively, indicating independent local curvatures with radii of 58, 66, and 39 ft.

### SECTION IV - SIGNIFICANCE OF RESULTS

The measurements reported herein have provided the first available indication of the actual degradation rate of a specular surface in a near-earth orbit and represent a significant advance over the estimates previously available for the design of solar energy collectors, thermal balance surfaces, etc. Repeated observation and analysis would undoubtedly narrow the present range of uncertainty. Together with the results from laboratory hypervelocity impact tests of very small particles on aluminized Mylar, these data should also prove useful in further defining the micrometeoroid environment at Echo I's orbital height, either with respect to the flux in the vicinity of the particle mass cutoff or the mass cutoff itself.

These measurements of Echo I can also serve as a cross check on measurements made by other methods of observation, such as by photoelectric or photographic photometry and high-resolution photography. With reference to the latter, Goodyear Aerospace with the cooperation of RCA Service Corporation personnel at the Kennedy Space Center, made some "quick-try" attempts in 1962 to obtain high-resolution optical and electro-optical photographs of Echo I, with uncertain results.<sup>7</sup> The tentative conclusion was that one (or more) specular highlights were being photographed repeatedly, and that Echo I was then still predominantly specular, with some macrotexture variation. This conclusion is entirely compatible with the recent and more definitive photometric indications.

Because Echo I (and now also Echo II) is continually immersed in the nearearth space environment and is continuing to integrate the effects of all the environmental parameters as encountered throughout the course of the earth's annual motion around the sun, it is increasing in its value both as a largescale test specimen in an existing ideal test for the technology of space materials and as a large-scale scientific probe of the near-earth micrometeoroid environment. It is suggested that Echo I and Echo II will continue to be sources of vital information for the astronautics engineer and space scientist and deserve the serious investigations that traditional astronomy would demand and pursue for natural satellites. Artificial satellites tend to become "naturalized citizens" of space.

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