

MASC

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FINAL REPORT

ON

THE DEVELOPMENT OF COMPUTATIONAL TECHNIQUES

FOR THE

IDENTIFICATION OF LINEAR AND NONLINEAR MECHANICAL SYSTEMS

SUBJECT TO RANDOM EXCITATION

Part II

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By

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A DYNAMICAL MOMENT TECHNIQUE
FOR THE
IDENTIFICATION OF SYSTEMS

I. Introduction

The purpose of this report is to describe a new identification technique that appears to be of superior practical significance due to its simplicity both in comprehending and applying the technique.

The technique is based upon relatively straightforward principles of random function analysis, merely involving the estimation of various order moments of the input and output processes.

In a preliminary investigation of the technique for identification of digitally simulated linear systems, the technique produced quite useful approximations to the actual system parameters, with a relatively small number of calculations. Hence, we feel that it warrants further investigation on this basis alone. Even more interesting is that the technique does not appear to be limited to linear systems alone. Since, theoretically, the technique is the same for linear or non-linear systems, that can be described by differential equations with polynomial non-linearities. Of course, it remains to be seen how applicable the technique will be for non-linear systems.

Before describing the technique, we wish to make two points clear relative to this technique and to the general subject of identification of real systems.

First, there exist techniques available for the study of linear systems. The newer techniques may employ cross-correlations or cross-spectral densities for the estimation of the impulse response function or for the frequency response function. These techniques require a random noise source and require the estimation of the functions involved at various time lags or at various frequencies. The techniques that have been used classically employ a simple sinusoidal driver at various frequencies to determine the frequency response of the system. Techniques, such as ours, that require only a parameter estimation are just now emerging. These techniques offer the promise of economy in calculations and time. The establishment of such techniques as practical tools will be of fundamental importance in future complex systems design and development.

References and explanations of the many approaches to this subject may be found in "Identification of Linear and Non-Linear Systems Cross-Correlations Techniques", by Dr. W. Gersch conducted by MASC for NASA-Goddard under NAS5-9741, October 1965.

The second point that must be made concerning the identification of systems is that the methods that have been developed, as well as the method we describe in this report, are methods

that will identify the analytical model or simulated model of the actual physical system. Hence, if the analytical or simulated model is not a satisfactory equivalent or approximation to the system, then clearly one is not identifying the real system.

Thus, any identification scheme is only as good as the analytical model that will be used to describe the physical system.

Of course, identification schemes can be used to help provide a better model to the system, if it is found that the original assumption is poor.

With this understanding of the proper role of identification techniques, we can now proceed to describe our approach.

II. A General Description of an Identification Technique for Linear Systems.

The basic principle of the technique that we propose is that the system undergoing study is time invariant and is being driven by a statistically stationary, non-white noise random process. This rules out flat wide band noise as inputs. We shall discuss the reason for this requirement below.

We shall assume, furthermore, that for all practical purposes the system is in steady state operation. Thus, the output process is a statistically stationary random process. This implies that all of its moments are constant in time.

Now if we write the general time invariant linear system subjected to random input as

$$\dot{\bar{y}}(t) = A \bar{y}(t) + \bar{x}(t), \quad (2.1)$$

where

$$\bar{y}(t) = \begin{pmatrix} y_1(t) \\ \vdots \\ y_n(t) \end{pmatrix}, \quad \bar{x}(t) = \begin{pmatrix} x_1(t) \\ \vdots \\ x_n(t) \end{pmatrix} \quad (2.2)$$

$$A = \begin{pmatrix} a_{11} & \dots & a_{1n} \\ \vdots & & \vdots \\ a_{n1} & \dots & a_{nn} \end{pmatrix},$$

then the \bar{y} -process is statistically stationary and, assuming that all moments of the \bar{x} -process exist, it follows that the moments

$$E \{ y_1^{r_1}(t) y_2^{r_2}(t) \dots y_n^{r_n}(t) x_1^{s_1}(t) \dots x_n^{s_n}(t) \} \quad (2.3)$$

exist and are constant in time.

Upon multiplying equation (2.1) by $\dot{\bar{y}}^T$ (the transpose of the state vector), we obtain

$$\dot{\bar{y}}(t) \bar{y}(t)^T = A \bar{y}(t) \bar{y}(t)^T + \bar{x}(t) \bar{y}(t)^T. \quad (2.4)$$

Taking expectations of the equation (2.4) yields

$$E \{ \dot{\bar{y}}(t) \bar{y}(t)^T \} = A E \{ \bar{y}(t) \bar{y}(t)^T \} + E \{ \bar{x}(t) \bar{y}(t)^T \} \quad (2.5)$$

or

$$[E \{ \dot{\bar{y}}(t) \bar{y}(t)^T \} - E \{ \bar{x}(t) \bar{y}(t)^T \}] E \{ \bar{y}(t) \bar{y}(t)^T \}^{-1} = A \quad (2.6)$$

assuming $E \{ \bar{y}(t) \bar{y}(t)^T \}^{-1}$ exists.

The relation (2.6) presents a general form of the solution of the general problem for the linear system.

It is easily seen that $E \{ \bar{y}(t) \bar{y}(t)^T \}$ is symmetric and, furthermore, since for any $(\alpha_1, \dots, \alpha_n)$,

$$E \{ \left| \sum_{i=1}^n \alpha_i y_i(t) \right|^2 \} \geq 0, \text{ it follows that } E \{ \bar{y}(t) \bar{y}(t)^T \}$$

is non-negative definite.

If the system equations are linearly independent, we shall expect that $E\{\bar{y}(t)\bar{y}^T(t)\}$ is positive definite and that its inverse exists. Thus, the matrix A is identified by estimation of the correlation matrices in Equation (2.6).

For our applications, we wish to be less general and somewhat more specific. We are assuming that the \bar{y} -process is a stationary, mean square differentiable process. Such a process is generated, for example, by passing a stationary mean square continuous process (i.e. a process whose covariance is continuous at the origin) through the time invariant linear system. This is the reason that we are ruling out wide band white noise processes. For, by passing a wide band, essentially white-noise, through the system the mean square differentiable property will cease to hold which, as can be shown, yields a bias to our estimations.

By our stationarity assumptions, it follows that the following moments, for the components of the vector \bar{y} -process,

$$E\{y_i^n(t)\} \quad \text{and} \quad E\{y_i^n(t)y_j^m(t)\} \quad (2.7)$$

exist and are constant in time.

Hence, it follows that

$$\frac{d}{dt} E\{y_i^n(t)\} = 0 \quad (2.8)$$

$$\frac{d}{dt} E\{y_i^n(t) y_j^m(t)\} = 0.$$

Due to our assumption of mean square differentiability, the derivatives in Equation (2.8) can be taken inside the expectations to give

$$a) E\{y_i^{n-1}(t) \dot{y}_i(t)\} = 0 \quad (2.9)$$

$$b) n E\{y_i^{n-1}(t) \dot{y}_i(t) y_j^m(t)\} + m E\{y_i^n(t) y_j^{m-1}(t) \dot{y}_j(t)\} = 0$$

In particular it follows that, for $n=m=1$ in 2.9b), $n=2$ in 2.9a)

$$a) E\{y_i(t) \dot{y}_i(t)\} = 0 \quad (2.10)$$

$$b) E\{y_i(t) \dot{y}_j(t)\} + E\{\dot{y}_i(t) y_j(t)\} = 0.$$

The first equality in Equation (2.10) states the well-known fact that a stationary process and its derivative are uncorrelated at any given time.

With these last few statements, we can now show how to identify a few specific linear systems.

III. Examples

In the following examples the x -process is a stationary mean square continuous, non-white, random process and the y -process is the stationary solution.

Example 1. A First-Order System

Consider the system

$$\dot{y}(t) = -a y(t) + x(t) \quad (3.1)$$

Upon multiplying Equation (3.1) by $y(t)$ and taking the expectations we have

$$E\{y(t) \dot{y}(t)\} = -a E\{y^2(t)\} + E\{y(t) x(t)\}. \quad (3.2)$$

But, by (2.10) it follows that

$$a = \frac{E\{y(t) x(t)\}}{E\{y^2(t)\}}. \quad (3.3)$$

Example 2. An Oscillator

Consider the second order system

$$\begin{aligned} a) \quad & \dot{y}_1(t) = y_2(t) \\ b) \quad & a \dot{y}_2(t) + b y_2(t) + c y_1(t) = x(t) \end{aligned} \quad (3.4)$$

Upon multiplying 3.4b) by $y_1(t)$, $y_2(t)$, $y_2^2(t)$ respectively and taking the expectations, we obtain

$$\begin{aligned} a E\{y_1(t) \dot{y}_2(t)\} + b E\{y_1(t) y_2(t)\} + c E\{y_1^2(t)\} &= E\{y_1(t) x(t)\} \\ a E\{y_2(t) \dot{y}_2(t)\} + b E\{y_2^2(t)\} + c E\{y_1(t) y_2(t)\} &= E\{y_2(t) x(t)\} \quad (3.5) \\ a E\{y_2^2(t) \dot{y}_2(t)\} + b E\{y_2^3(t)\} + c E\{y_2^2(t) y_1(t)\} &= E\{y_2^2(t) x(t)\} \end{aligned}$$

But, by equations (2.10) and (3.4a), we have

$$\begin{aligned} E\{y_2(t) \dot{y}_2(t)\} &= E\{y_2^2(t) \dot{y}_2(t)\} = 0 \\ E\{y_1(t) y_2(t)\} &= E\{y_1(t) \dot{y}_1(t)\} = 0 \quad (3.6) \\ E\{y_1(t) \dot{y}_2(t)\} &= -E\{\dot{y}_1(t) y_2(t)\} = -E\{y_2^2(t)\}. \end{aligned}$$

The equations 3.5) may now be written as

$$\begin{aligned} -a E\{y_2^2(t)\} &\quad + c E\{y_1^2(t)\} &= E\{y_1(t) x(t)\} \\ b E\{y_2^2(t)\} &&= E\{y_2(t) x(t)\} \quad (3.7) \\ b E\{y_2^3(t)\} &+ c E\{y_2^2(t) y_1(t)\} &= E\{y_2^2(t) x(t)\} \end{aligned}$$

from which a, b, c can be solved.

Example 3. A Coupled Oscillator

We now consider the system

$$\dot{y}_1(t) = y_2(t)$$

$$\begin{aligned}\dot{y}_2(t) &= -a y_2(t) - b y_1(t) + c[y_4(t) - y_2(t)] + d[y_3(t) - y_1(t)] \\ (3.8)\end{aligned}$$

$$\dot{y}_3(t) = y_4(t)$$

$$\dot{y}_4(t) = -c[y_4(t) - y_2(t)] - d[y_3(t) - y_1(t)] + x(t).$$

The second equation of (3.8) is multiplied by y_1, y_2 and the resulting equations are averaged. The fourth equation of (3.8) is multiplied by y_3, y_4 and the resulting equations are averaged.

On the basis of (2.10) and (3.8), the following four equations will result.

$$\begin{aligned}-E\{y_2^2\} &= -b E\{y_1^2\} + c E\{y_1 y_4\} + d[E\{y_1 y_3\} - E\{y_1^2\}] \\ 0 &= -a E\{y_2^2\} + c[E\{y_4 y_2\} - E\{y_2^2\}] + d E\{y_2 y_3\} \\ E\{y_4^2\} + E\{x y_3\} &= -c E\{y_2 y_3\} + d[E\{y_3^2\} - E\{y_1 y_3\}] \\ E\{x y_4\} &= +c[E\{y_4^2\} - E\{y_4 y_2\}] - d E\{y_1 y_4\}\end{aligned}\quad (3.9)$$

The set of four linear algebraic equations in four unknowns given by (3.9) are easily solved to determine the parameters (a, b, c, d) .

Thus, it is easily established that

$$c = \frac{\begin{vmatrix} E\{y_4^2\} + E\{xy_3\} & E\{y_3^2\} - E\{y_1y_3\} \\ E\{xy_4\} & -E\{y_1y_4\} \end{vmatrix}}{B},$$

$$d = \frac{\begin{vmatrix} -E\{y_3y_2\} & E\{y_4^2\} + E\{xy_3\} \\ E\{y_4^2\} - E\{y_4y_2\} & E\{xy_4\} \end{vmatrix}}{B}$$

where

$$B = \begin{vmatrix} -E\{y_3y_2\} & E\{y_3\} - E\{y_1y_3\} \\ E\{y_4^2\} - E\{y_4y_2\} & -E\{y_1y_4\} \end{vmatrix}$$

Similar expressions yield a, b , as well.

IV. Numerical Examples

In this section we present the results of parameter identification obtained by digitally simulating each of the examples above.

The systems digitally simulated are

a) $\dot{y}(t) = -y(t) + x(t)$

b)
$$\begin{cases} \dot{y}_1(t) = y_2(t) \\ \dot{y}_2(t) = -4y_2(t) - 16y_1(t) + x(t) \end{cases} \quad (4.1)$$

c)
$$\begin{cases} \dot{y}_1(t) = y_2(t) \\ \dot{y}_2(t) = -5y_2(t) - 100y_1(t) + x(t) \end{cases}$$

d)
$$\begin{cases} \dot{y}_1(t) = y_2(t) \\ \dot{y}_2(t) = -3y_2(t) - 9y_1(t) + 4[y_4(t) - y_2(t)] + 16[y_3(t) - y_1(t)] \\ \dot{y}_3(t) = y_4(t) \\ \dot{y}_4(t) = -4[y_4(t) - y_2(t)] - 16[y_3(t) - y_1(t)] + x(t) \end{cases}$$

The digital simulation of the input process, the $x(t)$ -process, is a stationary gaussian process generated by taking a moving average of the form

$$x_N = \sum_{i=0}^{K-1} c_i v_{N-i}, \quad (4.2)$$

$$N = 0, 1, 2, 3, \dots,$$

where the v's are independent identically distributed gaussian random variables generated by the usual computer routines, and the c's are weights chosen to obtain a specific covariance function.

For our example we chose K = 10,

$$c_i = 0.1 \quad \text{for } i = 0, 1, 2 \dots 9.$$

The moments calculated were simple arithmetic averages taken over the first 1300 output values, the second 1300 output values, and so on. The complete details of the digital simulation are given in Appendix A.

For the system,(4.1a), the first three groups of 1300 data points yield the estimates.

$E\{y^2\}$	$E\{xy\}$	a
5.6892	6.0682	1.0667
4.4014	4.8197	1.0950
3.3332	4.7634	1.0992

FIGURE 1.

For the system's (4.1 b,c),the first three groups of 1300 data points yielded (see following page),

	$E\{y_1^2\}$	$E\{y_1 y_2\}$	$E\{y_2^2\}$	$E\{y_1 x\}$	$E\{y_2 x\}$	a	b
System (4.1-b)	.0637	.0004	.4862	.5841	2.0500	4.2172	16.8006
	.0710	.0009	.5657	.6340	2.3313	4.1210	17.3080
	.0678	.0000	.5450	.5936	2.2500	4.1280	16.7810

System (4.1-c)	.0029	.0001	.0881	.2098	.4074	4.6190	103.72
	.0028	.0001	.0773	.2094	.3719	4.8120	103.27
	.0028	.0001	.0873	.2065	.4084	4.673	103.74

FIGURE 2

For system (4.1d), the correlation matrices and parameter estimates on the first two runs are

<u>Run I</u>	<u>y₁</u>	<u>y₂</u>	<u>y₃</u>	<u>y₄</u>
y ₁	.2253			
y ₂	.0000	.8223		
y ₃	.3216	.0839	.4700	
y ₄	-.0837	1.1044	.0004	1.6038
x	.1418	1.8412	.5172	3.5491

Parameter Estimates a=3.1790 b=9.194 c=4.3044 d=16.7258

Run II

y ₁	.2465			
y ₂	-.0004	.9255		
y ₃	.3505	.0924	.5105	
y ₄	-.0929	1.2505	.0004	1.8117
x	.1000	2.1328	.4837	3.9482

Parameter Estimates a=3.1708 b=9.2560 c=4.2525 d=16.8094

FIGURE 3

V. Comments and Conclusions

It is immediately obvious that the estimates obtained in the examples above are quite practical being less than 10%, and in most cases less than 5%, in error. The most significant feature of the estimates above is that their bias is almost always positive. That is, they do not fluctuate about the true value, but in fact they are almost all greater than the true value. We believe that this bias is a function of the nature of the noise input into the system. It can be demonstrated that if the noise input to the system approximates white noise into the system then the bias will increase. Hence, the noise source to the system should be smoothly filtered.

Exactly how the noise source effects the parameter estimates, remains to be studied and understood. It is our belief at this time, however, that a real system should be subjected to noise obtained from various filters to determine the robustness of the resulting system parameter estimates.

The stationarity assumption used, Equation (2.8), appears to be quite realistic in the simulated systems. Upon checking Figures 1, 2, 3, we see that the random functions and their derivatives possess correlation values that are orders of magnitude less than the other cross-correlations.

At this time we can only reiterate our sincere belief that this technique can be developed into one of significant engineering value both for linear and non-linear systems.

APPENDIX A

This section describes briefly the computer program used to simulate a system of five (or less) first order differential equations and to compute the second order moments to be used for calculating the parameters of the system as described in Section III.

The method of Runge-Kutta integration is used in Subroutine RK3 to simulate a system of first order differential equations which are supplied by Function Subroutines FN1, FN2, FN3, FN4, and FN5. The external excitation to the system is a filtered white Gaussian noise which is generated by Subroutines GENR, GAUSS, and RANDU. The filtering process used in the Subroutine GENR is a weighted sum of NF number of independent noise samples generated by Subroutines GAUSS and RANDU. The second order moments are estimated in Subroutine MOMENT, the method used there is to calculate the sample moments of N samples, that is, sample moment $m_{jk} = \frac{1}{N} \sum_{i=1}^N y_i(j) y_i(k)$.

Autocorrelation of the variables up to 9 lags are calculated in Subroutine AUTCOR in order to check the smoothness of the noise input as compared to system. As mentioned in Section II, the noise must have smooth correlation function in order for the method to be valid.

Different systems can be simulated by using different Function Subroutines FN1, FN2, FN3, FN4 and FN5. Different choices of the number of samples, the input noise level, the filter constants and the sampling interval are controlled by 7 input cards. The description and the formats of the 7 input cards are as follows.

	Col. No.	Format	Description
1st Card	1	I1	0 if the simulated sample points are not to be printed. 1 if the simulated sample points are to be printed.
	2 - 3	I2	Number of runs desired.
	4	I1	0 if each run starts from zero initial point. 1 if success runs are desired.
	1 - 10	I10	Any odd integer up to 9 digits for the first entry to Subroutine GAUSS.
2nd Card	11 - 20	F10.1	Standard deviation of the noise.
	21 - 30	F10.1	Mean value of the noise.
3rd Card	1 - 2	I2	Number of filter coefficients.
4th Card	1 - 80	16I5.1	Filter coefficients, as many cards as needed should be used.
5th Card	1 - 10	F10.1	Sample interval for integration.
6th Card	1 - 2	I2	Number of intervals between values of samples stored for calculation. 1 is used for the results in Section IV.
7th Card	1 - 4	I4	Number of sample points, N, desired for each run, $N \leq 1300$. If larger number is desired, the dimension of y in the Main Program has to be changed accordingly. Note that the average of the moments estimated from successive K runs is equivalent to the moments estimated from KxN number of samples.

The output consists of:

- A) The mean value of each variable.
- B) The second order moment matrix.
- C) The autocorrelation up to 9 lags for each variable.
- D) Sample points simulated if desired.

The prime reason for the simplicity of the program included in this appendix is that it was used mainly to check the moment estimates and establish the overall credibility and practicality of the technique for identifying simple simulated systems.

This program, therefore, must be considered as preliminary in that we have not allowed for an arbitrary number of degrees of freedom and we have not included the final arithmetic calculations of the estimated system parameters.

The studies included within this report are the initial investigations of the identification technique. Future investigations of this technique will include the composition of a comprehensive program that will accommodate linear systems of n-degrees of freedom and will carry the calculations to the final arithmetic stages providing a print out of the parameter estimates for prescribed classes of systems.

```

DIMENSION Y(6,1300),X(1300),Y1(6),FC(20),FMON(6),SMON(6),XMON(21)
1  FX(120),R(10),FN1,FN2,FN3,FN4,FN5,FN6
1  EXTERNAL FN1,FN2,FN3,FN4,FN5,FN6
READ(5,500) T0,NR,IR
READ(5,501) IX,SAN
READ(5,502) NFC,(I=1,NFC)
READ(5,503) STEP,KS,NV
KR=NR
DO 15 I=1,NFC
CALL GAUSS(IX,S,AM,V)
15  FX(1)=V
      TI=0
      DO 10 I=0,6
10   Y1(I)=0
      DO 17 J=1,NV
17   CALL GENR(IF,FC,NFC,FX,IX,S,AM)
20   X(J)=F
      CALL RK3(FN1,FN2,FN3,FN4,FN5,FN6,STEP,TI,Y1,KS,NV,X)
      DO 25 J=1,6
25   Y1(J)=Y(J,NV)
      TI=TI+NV*K$*STEP
      DO 30 J=1,NV
30   Y(6,J)=X(J)
      CALL MOMENT(Y,6,NV,1,FMON,SMON,XMON)
      WRITE(6,604)
      WRITE(6,602)(FMON(J),J=1,6)
      DO 40 J=1,6
40   CALL LOC(J,1,J1,6,1)
      CALL LOC(J,J,J2,6,1)
      WRITE(6,602)(XMON(K),K=J1,J2)
      DO 45 I=1,6
45   CALL AUTCOR(Y,I,6,NV,10,R)
      WRITE(6,605)(R(I),I=1,10)
45   IF(10)50,50
50   WRITE(6,601) ((Y(J,I),I=1,6),I=1,NV)
55   KR=KR-1
      IF(KR)100,100,60
60   IF(IR)17,5,17
500  FORMAT(11,12,11)
501  FORMAT(11,2F10,1)
502  FORMAT(12,16F5,1)
503  FORMAT(F10,1/12/14)
601  FORMAT(1H0,6E16,8)
602  FORMAT(1H1,7X,Y1,16X,'Y2',16X,'Y3',16X,'Y4',16X,'Y5',16X,'X')
604  FORMAT(1H0,10E11,3)
605  FORMAT(1H0,10E11,3)
100 STOP
END
SUBROUTINE AUTCOR(X,NOVAR,NV,ND,KS,R)
  DIMENSION X(1),R(1)
  DO 10 J=1,KS
10   R(J)=0.0
  NOKS=NO-KS+1
  DO 50 I=1,NOKS
50   IJ=(I-1)*NV+NOVAR
  DO 50 K=1,KS
50   IK=(I+K-2)*NV+NOVAR
  DO 60 K=1,KS
60   R(IK)=R(IK)+X(IJ)*X(IK)
  DO 60 K=1,KS
60   R(K)=R(K)/FLOAT(NOKS)
  RETURN
END
FUNCTION FN1(T,Y1,Y2,Y3,Y4,Y5,Y6,X)
FN1=-Y1+X
RETURN
END
FUNCTION FN2(T,Y1,Y2,Y3,Y4,Y5,Y6,X)
FN2=Y3
RETURN
END
FUNCTION FN3(T,Y1,Y2,Y3,Y4,Y5,Y6,X)
FN3=-4.0*Y3-16.0*Y2+X
RETURN
END
FUNCTION FN4(T,Y1,Y2,Y3,Y4,Y5,Y6,X)
FN4=Y5
RETURN
END
FUNCTION FN5(T,Y1,Y2,Y3,Y4,Y5,Y6,X)
FN5=-5.0*Y5-100.0*Y4+X
RETURN
END
FUNCTION FN6(T,Y1,Y2,Y3,Y4,Y5,Y6,X)
FN6=0.

```

END

SUBROUTINE RK3

PURPOSE
INTEGRATES A SYSTEM OF SIX FIRST ORDER DIFFERENTIAL EQUATIONS AND PRODUCES A TABLE OF INTEGRATED VALUES

USAGE
CALL RK3(FN1,FN2,FN3,FN4,FN5,FN6,H,XI,YI,K,N,VAL,F)

DESCRIPTION OF PARAMETERS
FN1-FN6 -SIX USER-SUPPLIED FUNCTION SUBPROGRAMS GIVING DY/DX AS A FUNCTION OF (X,Y1,Y2,Y3,Y4,Y5,Y6)
-STEP SIZE
-INITIAL VALUE OF X
-VECTOR OF LENGTH SIX CONTAINING THE INITIAL VALUES FOR Y1,Y2,Y3,Y4,Y5,Y6
K -THE DESIRED NUMBER OF STEPS OF SIZE H BETWEEN VALUES OF THE INTEGRALS STORED IN VAL
N -THE NUMBER OF VALUES TO BE RESTORED IN VAL.
-FINAL VALUE OF X WILL BE X+G(N*K*H)
VAL -RESULTANT SIX BY N MATRIX CONTAINING THE INTEGRATED VALUES OF THE SIX EQUATIONS
F - INPUT TO THE SYSTEM, ARRAY CF N

REMARKS
NONE

SUBROUTINES AND FUNCTION SUBPROGRAMS REQUIRED
FN1,FN2,FN3,FN4,FN5,FN6 AS NOTED ABOVE
CALLING PROGRAM MUST HAVE FORTRAN EXTERNAL STATEMENT
CONTAINING NAMES OF FUNCTION SUBPROGRAMS LISTED IN CALL TO RK3

METHOD
EXTENSION OF FOURTH ORDER RUNGE-KUTTA INTEGRATION ON A RECURSIVE BASIS AS SHOWN IN F.B. HILDEBRAND, 'INTRODUCTION TO NUMERICAL ANALYSIS', MCGRAW-HILL, NEW YORK, 1956

SUBROUTINE RK3(FN1,FN2,FN3,FN4,FN5,FN6,H,XI,YI,K,N,VAL,F)
DIMENSION Y(6),Y(6),S1(6),S2(6),S3(6),S4(6),VAL(1),F(1)

IF A DOUBLE PRECISION VERSION OF THIS ROUTINE IS DESIRED, THE COLUMN 1 SHOULD BE REMOVED FROM THE DOUBLE PRECISION STATEMENT WHICH FOLLOWS.

DOUBLE PRECISION H,XI,YI,VAL,Y,S1,S2,S3,S4,H2,X,T,XA,A1,A2,A3,
A4,A5,A6,FN1,FN2,FN3,FN4,FN5,FN6

THE C MUST ALSO BE REMOVED FROM DOUBLE PRECISION STATEMENTS APPEARING IN OTHER ROUTINES USED IN CONJUNCTION WITH THIS ROUTINE.

USER FUNCTION SUBPROGRAMS FN1-FN6 MUST BE IN DOUBLE PRECISION

H2=H/2.
DO 10 I=1,6
Y(I)=Y(I)
DO 70 L=L-1,N
L=(L-1)*6
DO 69 JJ=1,K

COMPUTE K SUB 0

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DO 30 I=1,6
GO TO (2,1,22,23,24,25,26) I
T=FN1(X,Y(1),Y(2),Y(3),Y(4),Y(5),Y(6),F(LL))
Y(1)=Y(1)
DO 70 L=L-1,N
L=(L-1)*6
DO 69 JJ=1,K
H2=H/2.
COMPUTE K SUB 0
DO 30 I=1,6
GO TO (2,1,22,23,24,25,26) I
T=FN1(X,Y(1),Y(2),Y(3),Y(4),Y(5),Y(6),F(LL))
T=FN2(X,Y(1),Y(2),Y(3),Y(4),Y(5),Y(6),F(LL))
T=FN3(X,Y(1),Y(2),Y(3),Y(4),Y(5),Y(6),F(LL))
T=FN4(X,Y(1),Y(2),Y(3),Y(4),Y(5),Y(6),F(LL))
T=FN5(X,Y(1),Y(2),Y(3),Y(4),Y(5),Y(6),F(LL))
T=FN6(X,Y(1),Y(2),Y(3),Y(4),Y(5),Y(6),F(LL))

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30 S1(I)=H*1
XA=XEH2
A1=Y(1)*S1(I)/2.
A2=Y(2)*S1(I)/2.
A3=Y(3)*S1(I)/2.
A4=Y(4)*S1(I)/2.
A5=Y(5)*S1(I)/2.
A6=Y(6)*S1(I)/2.

C COMPUTE K SUB 1
DO 40 I=1,6
GO TO (31,32,33,34,35,36),I
31 T=FN1(XA,A1,A2,A3,A4,A5,A6,F(LL))
32 T=FN2(XA,A1,A2,A3,A4,A5,A6,F(LL))
33 T=FN3(XA,A1,A2,A3,A4,A5,A6,F(LL))
34 T=FN4(XA,A1,A2,A3,A4,A5,A6,F(LL))
35 T=FN5(XA,A1,A2,A3,A4,A5,A6,F(LL))
36 T=FN6(XA,A1,A2,A3,A4,A5,A6,F(LL))
40 S2(I)=H#1
A1=Y(1)*S2(I)/2.
A2=Y(2)*S2(I)/2.
A3=Y(3)*S2(I)/2.
A4=Y(4)*S2(I)/2.
A5=Y(5)*S2(I)/2.
A6=Y(6)*S2(I)/2.

C COMPUTE K SUB 2
DO 50 I=1,6
GO TO (41,42,43,44,45,46),I
41 T=FN1(XA,A1,A2,A3,A4,A5,A6,F(LL))
42 T=FN2(XA,A1,A2,A3,A4,A5,A6,F(LL))
43 T=FN3(XA,A1,A2,A3,A4,A5,A6,F(LL))
44 T=FN4(XA,A1,A2,A3,A4,A5,A6,F(LL))
45 T=FN5(XA,A1,A2,A3,A4,A5,A6,F(LL))
46 T=FN6(XA,A1,A2,A3,A4,A5,A6,F(LL))
50 S3(I)=H#1
XA=XEH
A1=Y(1)*S3(I)
A2=Y(2)*S3(I)
A3=Y(3)*S3(I)
A4=Y(4)*S3(I)
A5=Y(5)*S3(I)
A6=Y(6)*S3(I)

C COMPUTE K SUB 3
DO 65 I=1,6
GO TO (51,52,53,54,55,56),I
51 T=FN1(XA,A1,A2,A3,A4,A5,A6,F(LL))
52 T=FN2(XA,A1,A2,A3,A4,A5,A6,F(LL))
53 T=FN3(XA,A1,A2,A3,A4,A5,A6,F(LL))
54 T=FN4(XA,A1,A2,A3,A4,A5,A6,F(LL))
55 T=FN5(XA,A1,A2,A3,A4,A5,A6,F(LL))
56 T=FN6(XA,A1,A2,A3,A4,A5,A6,F(LL))
65 S4(I)=H#1
XA=XEH

C COMPUTE NEW VALUES OF INTEGRALS
DO 69 I=1,6
Y(I)=Y(I)*S1(I)*2.*S2(I)*2.*S3(I)*S4(I)/6.
DO 70 I=1,6
II=IL
VAL(II)=Y(II)
RETURN
END
SUBROUTINE GENR(F,FC,NFC,FX,IY,S,AM)
DIMENSION FC(1),FX(1)
L=NFC-1
DO 10 J=1,L

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10 FX(K+1)=FX(K)
CALL GAUSS(IX,S,AM,V)
FX(1)=V
F=0
DO 20 I=1,NFC
   F=F+F(X(I))*FX(I)
20 RETURN
END

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SUBROUTINE GAUSS

PURPOSE COMPUTES A NORMALLY DISTRIBUTED RANDOM NUMBER WITH A GIVEN MEAN AND STANDARD DEVIATION

GAUSS001
GAUSS002
GAUSS003
GAUSS004
GAUSS005
GAUSS006
GAUSS007
GAUSS008

PURPOSE COMPUTES A NORMALLY DISTRIBUTED RANDOM NUMBER WITH A GIVEN MEAN AND STANDARD DEVIATION

USAGE CALL GAUSS(IX,S,AM,V)
 DESCRIPTION OF PARAMETERS
 IX - IX MUST CONTAIN AN ODD INTEGER NUMBER WITH NINE OR
 LESS DIGITS ON THE FIRST ENTRY TO GAUSS. THEREAFTER
 IT WILL CONTAIN A UNIFORMLY DISTRIBUTED INTEGER RANDOM
 NUMBER GENERATED BY THE SUBROUTINE FOR USE ON THE NEXT
 ENTRY TO THE SUBROUTINE.
 S - THE DESIRED STANDARD DEVIATION OF THE NORMAL
 DISTRIBUTION.
 AM - THE DESIRED MEAN OF THE NORMAL DISTRIBUTION.
 V - THE VALUE OF THE COMPUTED NORMAL RANDOM VARIABLE

REMARKS THIS SUBROUTINE USES RANDU WHICH IS MACHINE SPECIFIC
SUBROUTINES AND FUNCTION SUBPROGRAMS REQUIRED
RANDU

METHOD USES 12 UNIFORM RANDOM NUMBERS TO COMPUTE NORMAL RANDOM NUMBERS BY CENTRAL LIMIT THEOREM. THE RESULT IS THEN ADJUSTED TO MATCH THE GIVEN MEAN AND STANDARD DEVIATION. THE UNIFORM RANDOM NUMBERS COMPUTED WITHIN THE SUBROUTINE ARE FOUND BY THE POWER RESIDUE METHOD.

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SUBROUTINE GAUSS(IX,S,AM,V)
A=0
DO 50 I=1,12
CALL RANDU(IX,IY,Y)
IX=IY
A=A*Y
V=(A-6.0)*S&AM
50 RETURN

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SUBROUTINE RANDU

PURPOSE COMPUTES UNIFORMLY DISTRIBUTED RANDOM REAL NUMBERS BETWEEN 0 AND 1.0 AND RANDOM INTEGERS BETWEEN 2**31. EACH ENTRY USES AS INPUT AN INTEGER AND PRODUCES A NEW INTEGER AND RANDOM NUMBER.

USAGE CALL RANDU(IX,IY,YFL)

DESCRIPTION OF PARAMETERS

IX - FOR THE FIRST ENTRY THIS MUST CONTAIN ANY ODD INTEGER NUMBER WITH NINE OR LESS DIGITS. AFTER THE FIRST ENTRY IX SHOULD BE THE PREVIOUS VALUE OF IY COMPUTED BY THIS SUBROUTINE.

IY - A RESULTANT INTEGER RANDOM NUMBER REQUIRED FOR THE NEXT ENTRY TO THIS SUBROUTINE. THE RANGE OF THIS NUMBER IS BETWEEN ZERO AND $2^{31} - 1$.

REMARKS THIS SUBROUTINE IS SPECIFIC TO SYSTEM/360
 THIS SUBROUTINE WILL PRODUCE 2**29 TERMS
 BEFORE REPEATING

SUBROUTINES AND FUNCTION SUBPROGRAMS REQUIRED
NINE

C METHOD
C POWER RESIDUE METHOD DISCUSSED IN IBM MANUAL C20-8011,
C RANDOM NUMBER GENERATION AND TESTING

C SUBROUTINE RANDU(IX,IY,YFL)
C IY=IX*65539
C IF(IY)5 6
C 5 IY=IY*2147483647&1
C 6 YFL=IY
C YFL=YFL*.4656613E-9
C RETURN
C END

C SUBROUTINE MOMENT(A,NV,NO,IXM,FMON,SMON,XMON)

A- DATA MATRIX NV BY NO
NV- NO OF VARIABLES
IXM- INDEX FOR CROSS MOMENTS

I XM=1 WANT CROSS MOMENTS
IXM=0 DO NOT WANT CROSS MOMENTS

FMON- FIRST MOMENTS ARRAY OF NV
SMON- SECOND MOMENTS ARRAY OF NV
CROSS MOMENTS ARRAY OF (NV*(NV+1))/2

DIMENSION A(1),FMON(1),SMON(1),XMON(1)

INITIALIZATION

DO 10 J=1,NV
FMON(J)=0
10 SMON(J)=0
DO 20 J=1,NV
DO 20 L=1,J
CALL LOC(J,L,K,NV,NV,1)
20 XMON(K)=0

C CALCULATE FIRST MOMENTS

DO 30 I=1,NO
DO 30 J=1,NV
IJ=(I-1)*NV+J
30 FMON(J)=FMON(J)+A(IJ)
DO 40 J=1,NV
40 FMON(J)=FMON(J)/NO

C CALCULATE SECOND MOMENTS

IF(IXM)60,45,60
45 DO 50 I=1,NO
DO 50 J=1,NV
IJ=(I-1)*NV+J
50 SMON(J)=SMON(J)+A(IJ)*A(IJ)
DO 55 J=1,NV
55 SMON(J)=SMON(J)/NO
GO TO 100

C CALCULATE CROSS MOMENTS

60 DO 70 I=1,NO
DO 70 J=1,NV
DO 70 L=1,J
CALL LOC(J,L,K,NV,NV,1)
IJ=(I-1)*NV+J
IL=(I-1)*NV+L
70 XMON(K)=A(IJ)*A(IL)+XMON(K)
KN=(NV*(NV+1))/2
DO 75 K=1,KN
75 XMON(K)=XMON(K)/NO
100 RETURN

SUBROUTINE LOC

PURPOSE COMPUTE A VECTOR SUBSCRIPT FOR AN ELEMENT IN A MATRIX OF SPECIFIED STORAGE MODE

USAGE CALL LOC (I, J, IR, N, M, MS)

DESCRIPTION OF PARAMETERS

I	- ROW NUMBER OF ELEMENT
J	- COLUMN NUMBER OF ELEMENT
IR	- RESULTANT VECTOR SUBSCRIPT
N	- NUMBER OF ROWS IN MATRIX
M	- NUMBER OF COLUMNS IN MATRIX
MS	- ONE DIGIT NUMBER FOR STORAGE MODE OF MATRIX
0	- GENERAL
1	- SYMMETRIC
2	- DIAGONAL

REMARKS
NONE

SUBROUTINES AND FUNCTION SUBPROGRAMS REQUIRED

METHOD

MS=0	SUBSCRIPT IS COMPUTED FOR A MATRIX WITH N*M ELEMENTS IN STORAGE (GENERAL MATRIX)
MS=1	SUBSCRIPT IS COMPUTED FOR A MATRIX WITH N*(N+1)/2 IN STORAGE (UPPER TRIANGULAR OF SYMMETRIC MATRIX). IF ELEMENT IS IN LOWER TRIANGULAR PORTION, SUBSCRIPT IS CORRESPONDING ELEMENT IN UPPER TRIANGLE.
MS=2	SUBSCRIPT IS COMPUTED FOR A MATRIX WITH N ELEMENTS IN STORAGE (DIAGONAL ELEMENTS OF DIAGONAL MATRIX). IF ELEMENT IS NOT ON DIAGONAL (AND THEREFORE NOT IN STORAGE), IR IS SET TO ZERO.

SUBROUTINE LOC(I,J,IR,N,M,MS)

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C
I X = I
J X = J
10 IF(MS=1) 10,20,30
10   GO TO 36
20 IF((IX-JX) 22,24,24
22 IX=IX&(JX*JX-JX)/2
22   GO TO 36
24 IX=JX&(IX*IX-IX)/2
24   GO TO 36
30 IX=0
30   IF(IX-JX) 36,32,36
32 IX=IX
36 IX=IRX
      RETURN
END

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