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ANALOGIES BETWEEN EM AND ACOUSTIC WAVES*

By

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Introduction

The purpose of this report is to summarize the results of a considerable amount of work in the field of acoustic simulation of radar return so that the reader may be readily confident about the validity of this inexpensive laboratory tool.

Acoustic simulation readily permits experimental verification of scatter theories which would otherwise be costly and time consuming.

Wave Equations

Electromagnetic

Acoustic

| | |
|---|--|
| $\nabla \times E = -\mu \frac{\partial H}{\partial t}$ $\nabla \times H = \epsilon \frac{\partial E}{\partial t}$ | $\nabla P = -\rho_l \frac{\partial u}{\partial t}$ $\nabla \cdot u = -K_l \frac{\partial P}{\partial t}$ |
|---|--|

for plane sinusoidal waves in the x-direction:

| | |
|--|--|
| $\frac{dEy}{dx} = -i\omega\mu Hz$ $\frac{dHz}{dx} = -i\omega\epsilon Ey$ | $\frac{dP}{dx} = -i\omega\rho_l un$ $\frac{du_x}{dx} = -i\omega K_l P$ |
|--|--|

taking the divergence of (1) and the substitution of (2)

| | |
|---|---|
| $\nabla^2 E = \mu\epsilon \frac{\partial^2 E}{\partial t^2}$ $\nabla^2 H = \mu\epsilon \frac{\partial^2 H}{\partial t^2}$ | $\nabla^2 P = \rho_l K_l \frac{\partial^2 P}{\partial t^2}$ $\nabla^2 u = \rho_l K_l \frac{\partial^2 u}{\partial t^2}$ |
|---|---|

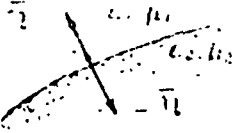
where

| | |
|----------------------------------|-----------------------------------|
| E = electric field vector | P = pressure field scalar |
| H = magnetic field vector | u = particle velocity vector |
| ϵ = dielectric constant | ρ_l = density of the medium |
| μ = permeability | K_l = compressibility of medium |

Simulation of electromagnetic waves using acoustic waves is always feasible when their respective boundary conditions are nearly the same.

BOUNDARY CONDITIONS

Electromagnetic



Most General Form

$$\bar{n} \times (\bar{E}_1 - \bar{E}_2) = Q \quad (1)$$

$$\bar{n} \times (\bar{H}_1 - \bar{H}_2) = J_s \quad (2)$$

When:

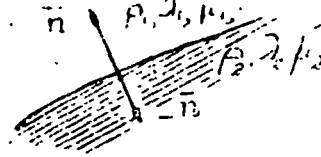
$$M_1/M_2 = \text{Dielectric/Dielectric} \quad Q = 0 \\ J_s = 0$$

$$M_1/M_2 = \text{Dielectric/Perf. Cond.} \quad Q \neq 0 \\ \sigma = \infty \quad \bar{E}_2 = \bar{H}_2 = 0 \quad J_s = 0$$

$$M_1/M_2 = \text{Dielectric/Imp. Cond.} \quad Q \neq 0 \\ \sigma = \text{finite} \quad J_s = 0$$

$$\text{Perfect} \\ M_1/M_2 = \text{Conductor/Imp. Cond.} \quad Q \neq 0 \\ \bar{E}_1 = \bar{H}_1 = 0 \quad J_s \neq 0$$

Acoustic



Most General Form

$$P_1 = P_2 \quad (1)$$

$$\bar{n} \cdot (\bar{U}_1 - \bar{U}_2) = 0 \quad (2)$$

When:

$$M_1/M_2 = \text{Liquid/Liquid} \quad P_1 = P_2 \\ \bar{n} \cdot (\bar{U}_1 - \bar{U}_2) = 0$$

$$M_1/M_2 = \text{Liquid/Solid Rigid} \quad P_1 = P_2 = 0 \\ \bar{n} \cdot (\bar{U}_1 - \bar{U}_2) = 0$$

$$M_1/M_2 = \text{Liquid/Semielastic Solid} \quad P_1 = P_2 = 0 \\ U_{\text{total}} = U_t = 2U_{\text{incident}}$$

$$M_1/M_2 = \text{Solid/Elastic Rigid Solid} \quad P_1 = P_2 = 0 \\ \bar{n} \cdot (\bar{U}_1 - \bar{U}_2) = 0$$

Conductors

Perfect Conductor: $\sigma \rightarrow \infty$

Imperfect Conductor: $\sigma = \text{finite}$

Insulator: $\sigma = 0$

Solid Surfaces

Rigid Surface: $\mu \Rightarrow \infty$

Elastic Surface: $\mu = \text{finite}$

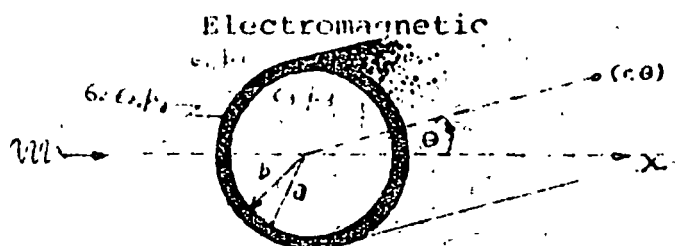
Liquid Surface: $\mu = 0$

For the EM and Acoustic field \bar{n} is a unit vector normal to the surface. The equations state that the tangential components of \bar{E} and \bar{H} are continuous across the boundary as are the pressure P and the normal component of the particle velocity \bar{v} .

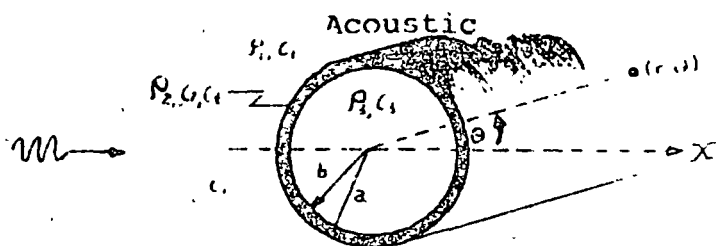
Boundary conditions at a perfect conductor require the tangential electric field to be zero and the tangential magnetic field to be two times the incident field. On the other hand at a perfectly elastic wall (pressure release surface) the dynamic pressure is zero and the normal component of the total particle velocity is twice the normal component of the incident particle velocity. In each of these situations there is no wave propagation beyond the interface.

When a plane electromagnetic wave is normally incident on a perfectly conducting plane surface, the field components are both parallel to the surface and the electric field component can be made equivalent to either the pressure of the particle velocity in the acoustic wave using a perfectly elastic boundary in the first case and a perfectly rigid boundary in the second.

SCATTERING FROM A CYLINDER



- $b = 0$ Dielectric Cylinder
(W. L. Weeks, 1964)
- $\sigma \rightarrow \infty$ Conducting Cylinder
(W. L. Weeks, 1964)
- $b \approx a$ Thin Cylindrical Shell



- $b = 0$ Solid Elastic Cylinder
(R. D. Doolittle, 1966)
- $\mu \rightarrow \infty$ Rigid Cylinder
- $b = 0, \mu = 0, \lambda = 0$ Liquid Cylinder
(P. Tamarkin, 1949)

$$\Phi_{\text{total}} = \Phi_{\text{incident}} + \Phi_{\text{scattered}}$$

$$\Phi_{\text{incident}} = \Phi_0 e^{iKx} = \Phi_0 \sum_{n=0}^{\infty} i^n \epsilon_n J_n(Kr) \cos(n\theta)$$

$$\Phi_{\text{scattered}} = \Phi_0 \sum_{n=0}^{\infty} i^n \epsilon_n a_n H_n^{(1)}(Kr) \cos(n\theta)$$

where: $\epsilon_n = 2 - \delta_{n0} = \begin{cases} 1 & \text{for } n = 0 \\ 2 & \text{for } n \neq 0 \end{cases}$

a_n = constant depending on the boundary conditions

J_n, H_n = Bessel and Hankel functions

$e^{-i\omega t}$ = time dependence has been suppressed

$\Phi = E$ = Electric Potential

σ = Conductivity

ϵ = Permittivity

μ = Permeability

$\Phi = P$ = Pressure Potential

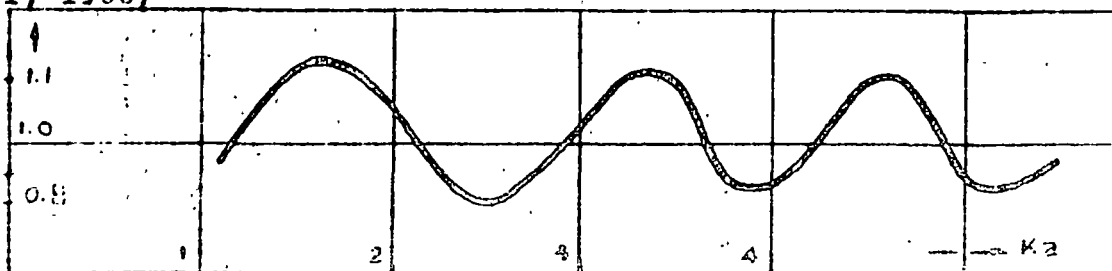
μ = Lamé Constant

ρ = density

$c_{1,t}$ = Velocities (Longitudinal, transverse)

(H. Uberal, 1966)

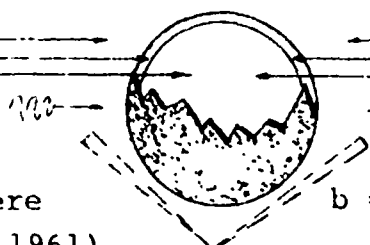
Scattering from Rigid or Conducting Cylinder



SCATTERING FROM A SPHERE

Electromagnetic

ϵ_1, μ_1, k_1
 ϵ_2, μ_2, k_2
 ϵ_3, μ_3



Acoustic

ρ_1, c_1
 ρ_2, c_2, c_t
 ρ_3, c_3

$b = 0$ Dielectric Sphere
(D. T. Thomas, 1961)

$\sigma \rightarrow \infty$ Conducting Sphere
(R. F. Goodrich, 1961)

$b \equiv a$ Spherical Shell
(A. L. Laden, 1952)

$b = 0$ Solid Elastic Sphere
(R. Hickling, 1962)

$\mu \rightarrow \infty$ Rigid Sphere
(R. B. Lindsay, 1960)

$b = 0, \mu = 0, \lambda \neq 0$ Liquid Sphere
(A. L. Laden, 1952)

$$\Phi_{\text{total}} = \Phi_{\text{incident}} + \Phi_{\text{scattered}}$$

$$\Phi_{\text{incident}} = \Phi_0 \exp(iKr \cos \theta) = \Phi_0 \sum_{n=0}^{\infty} (2n+1) i^n P_n(\cos \theta) j_n(Kr)$$

$$\Phi_{\text{scattered}} = \Phi_0 \sum_{n=0}^{\infty} a_n h_n(Kr) P_n(\cos \theta)$$

where a_n = constant depending on the boundary conditions

P_n = Legendre polynomial

h_n = spherical Bessel function

J_n = Bessel function

$e^{-i\omega t}$ = time dependence has been suppressed

$\Phi = E$ = Electric Potential

σ = Conductivity

ϵ = Permittivity

μ = Permeability

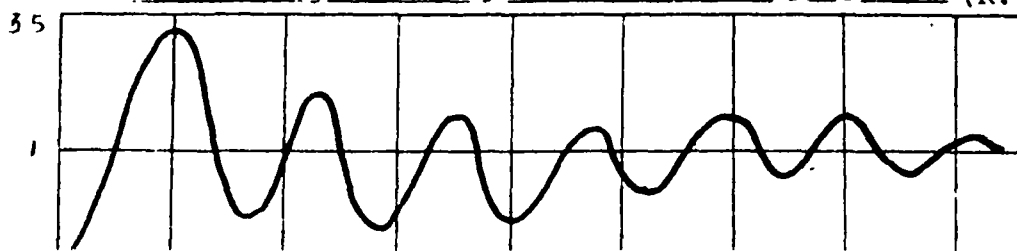
$\Phi = P$ = Pressure

μ = Lamé Const.

ρ = Density

$c_{1,t}$ = Velocities of Sound

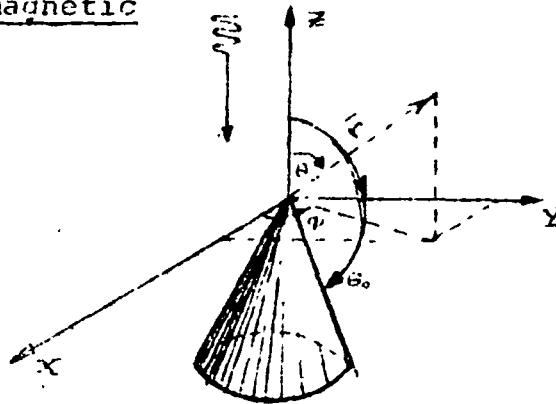
Scattering from Rigid or Conducting Sphere (R. F. Goodrich, 1961)



SCATTERING FROM A CONE

Electromagnetic

Acoustic

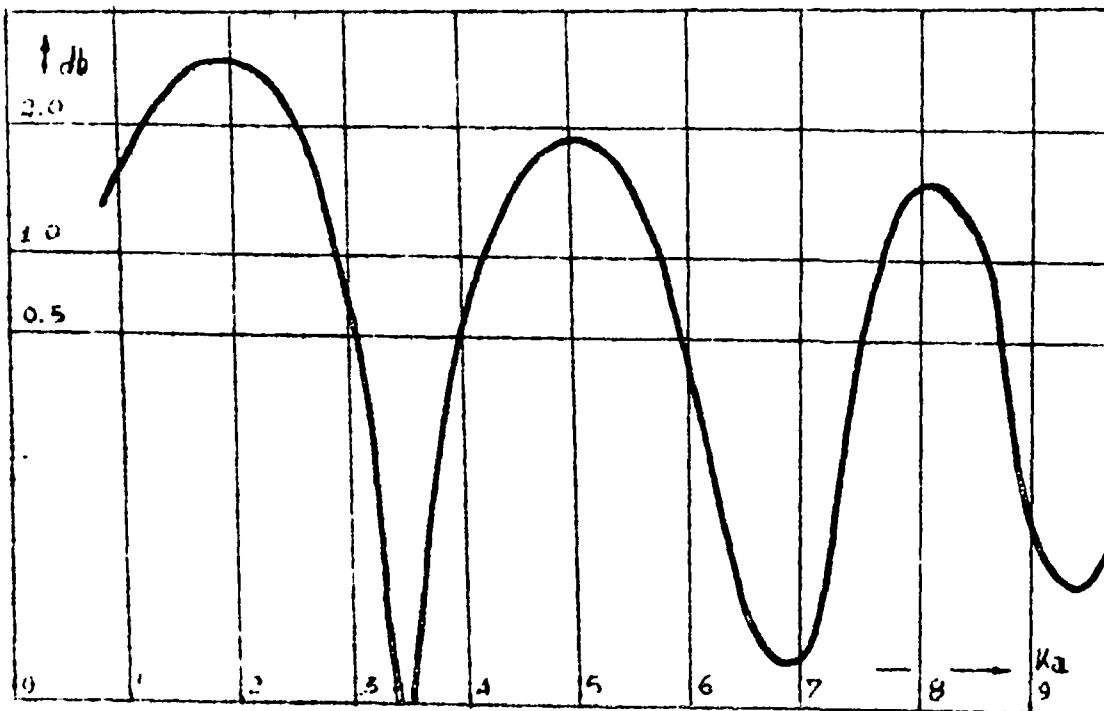


$$\begin{aligned} x &= r \cos \phi \sin \theta \\ y &= r \sin \phi \sin \theta \\ z &= r \cos \theta \end{aligned}$$

For the special case of a finite cone, flat-backed (base is circular or radius a), and a given height h with a semivertex angle $\pi - \theta_0$, the scattered far field amplitude is given by the following approximation formula (good for high frequencies). (R. F. Kleinman, 1966)

$$s = \frac{4i\pi}{\lambda^2} \tan^2 \theta_0 \int_0^h e^{2iKu} du$$

Flat-backed cone $\pi - \theta_0 = 4^\circ$ (R. F. Kleinman, 1966)



Experimental Verification



For nearly smooth surfaces, the surface characteristic constant $1/B = 0$, and (3.15) gives the scattering coefficient. This result compares very closely with published results [Nielson, 1960] for new ice as shown in table 2.

For rough (not nearly smooth) surfaces, (3.14) describes the relationship of the scattering coefficient σ_0 and other variables such as the angle of incidence θ , wavelength λ , standard deviation σ and surface covariance constant B , etc. Two curves of the scattering coefficient σ_0 versus θ for each of the three values of λ/B , 0.1, 0.5, and 1.0 for σ/λ equal to 0.05, and 0.1 are shown in figure 4. It may be noticed that as the surface becomes rougher, or as λ/B increases for a specified λ , the scattering coefficient curve becomes flatter, showing the relative importance of the contribution of the power return from the surface at angles other than those near zero. As expected, when the surface becomes smoother or $1/B$ decreases, the received power seems to come primarily from near-zero angles. These curves are quite similar to those recently published [Campbell, 1959; Dye, 1959; Edison, 1960]. The experimental data [Nielson, 1960] on desert and new ice also seems to follow the pattern of these theoretical curves described above.

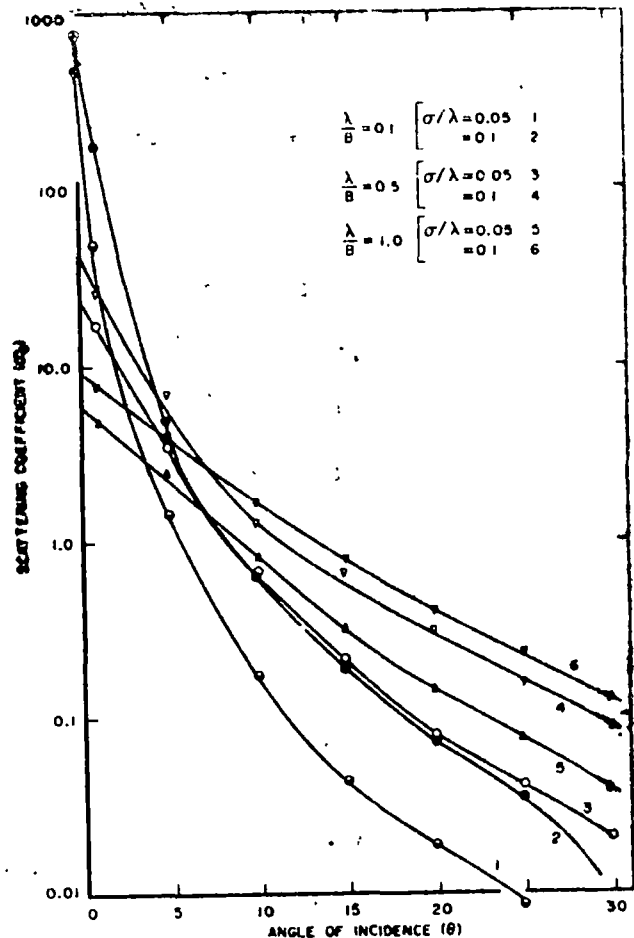


FIGURE 4. Scattering coefficient versus the angle of incidence.

The scattering coefficient (σ_0) for nearly smooth surfaces is inversely proportional to the wavelength, but varies directly with (σ^2) , $(\theta \cot^2 \theta)$ and $1/B$, where σ , θ , B are standard deviation, angle of incidence, and the terrain characteristic constant respectively. For rough surfaces it has a negative exponential factor, where the exponent is made up of $\frac{\sigma^2 \cos^2 \theta}{\lambda^2}$

times a constant. The surface characteristic constants B and σ can be calculated from the radar return data. Although approximate, the theoretical results agree well with the experimental data; and therefore, suggest the usefulness of the approach. The application of these results may be extended to the moon-echo data, with proper corrections for Faraday and liberation effects, etc. This investigation has established that for near-vertical incidence, the normalized autocovariance for the terrain elevation is more often of the exponential form $\exp(-r/B)$ rather than the Gaussian form, $\exp(-r^2/B)$. The former may well be more appropriate for finer terrain irregularities than those considered in this

Comparison of theoretical versus experimental scattering coefficient (Normalized)

| θ° | σ_0 Theoretical | σ_0 Experimental |
|----------------|------------------------|-------------------------|
| 30 | 1.000 | 1.000 |
| 40 | 0.251 | 0.308 |
| 50 | 0.084 | 0.089 |
| 60 | 0.022 | 0.024 |
| 70 | 0.004 | 0.004 |

* H. S. HAYRE & R. K. MOORE. Theoretical Scattering Coef for near Vertical Incidence from Contour Maps, Journal of Research Vol. B5D, Sept-1961 pp 431.

The distinction between beam-width and pulse-length limiting of the illuminated area is a function of altitude¹ and is given by

$$1 + \frac{v\tau}{2h} = \sec \theta_0 \quad (6-1)$$

where

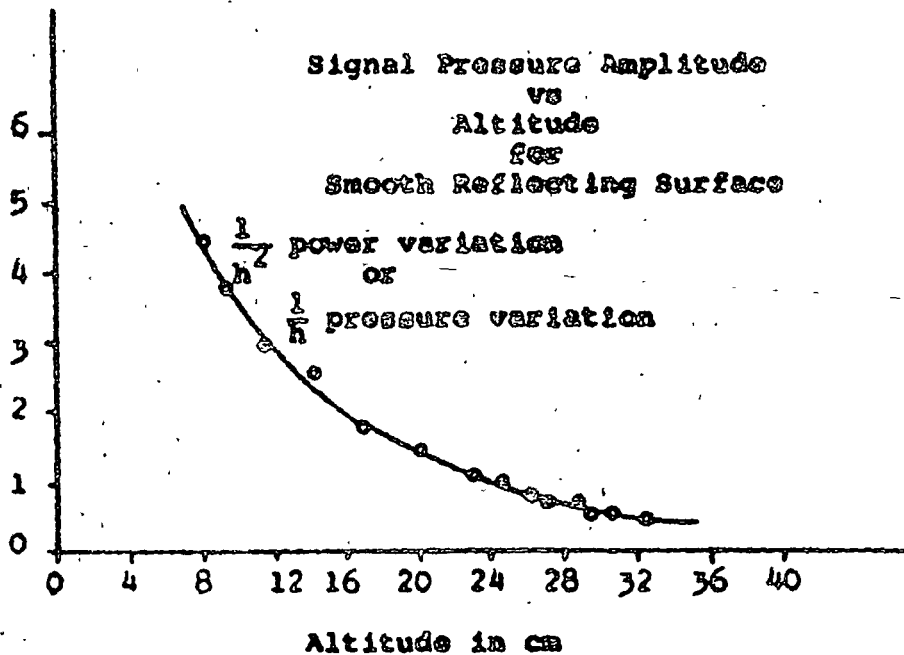
v is the velocity of propagation,

τ is the pulse width,

h is the altitude, and

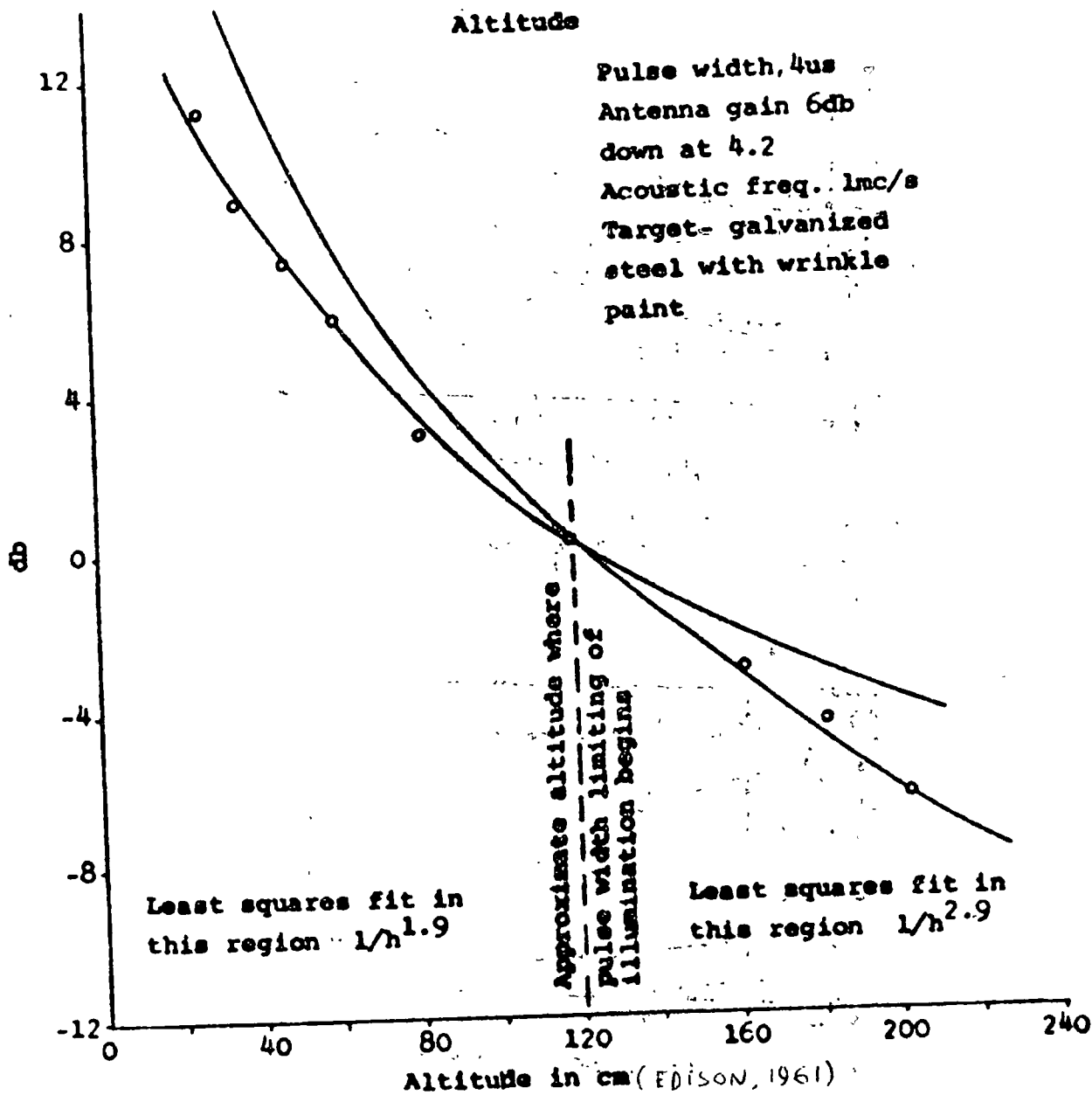
θ_0 is the effective half-angle of the antenna pattern.

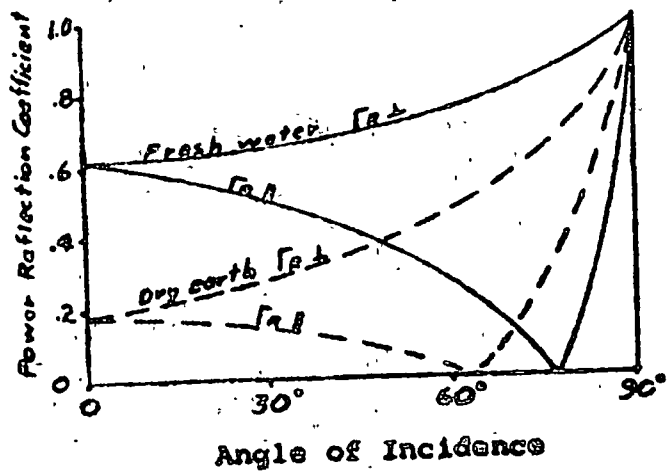
Relative Mean Pressure Amplitude



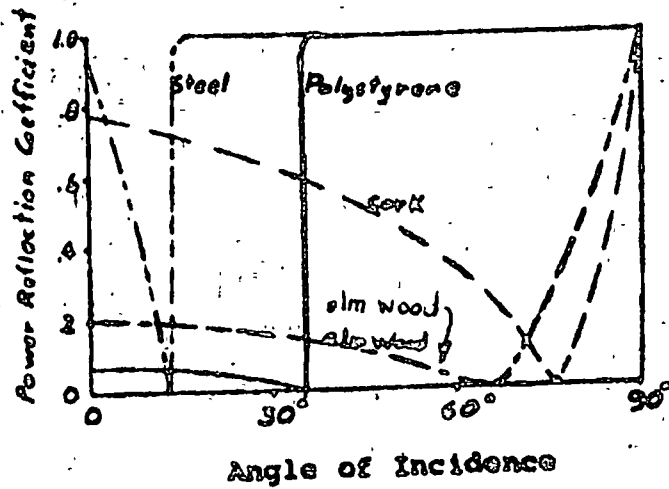
Signal vs Altitude over Smooth Surface (EDISON, 1961)

Mean Scatter-Power Signal
vs
Altitude

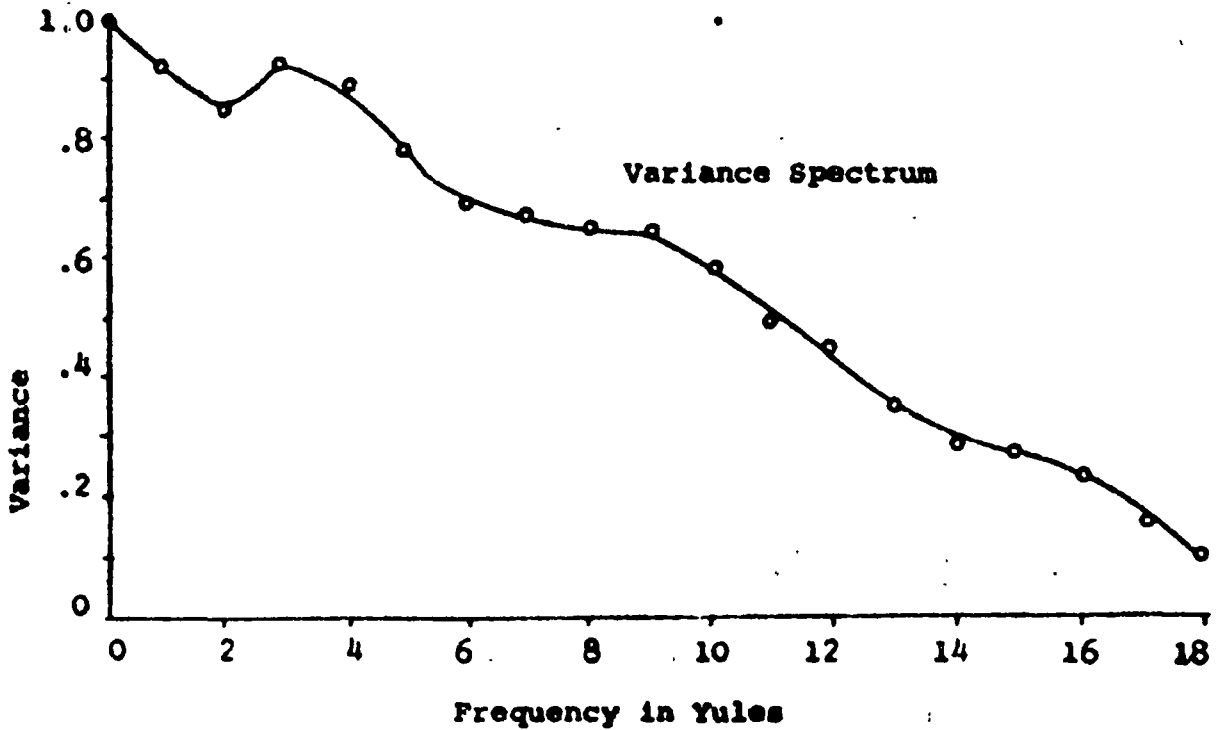
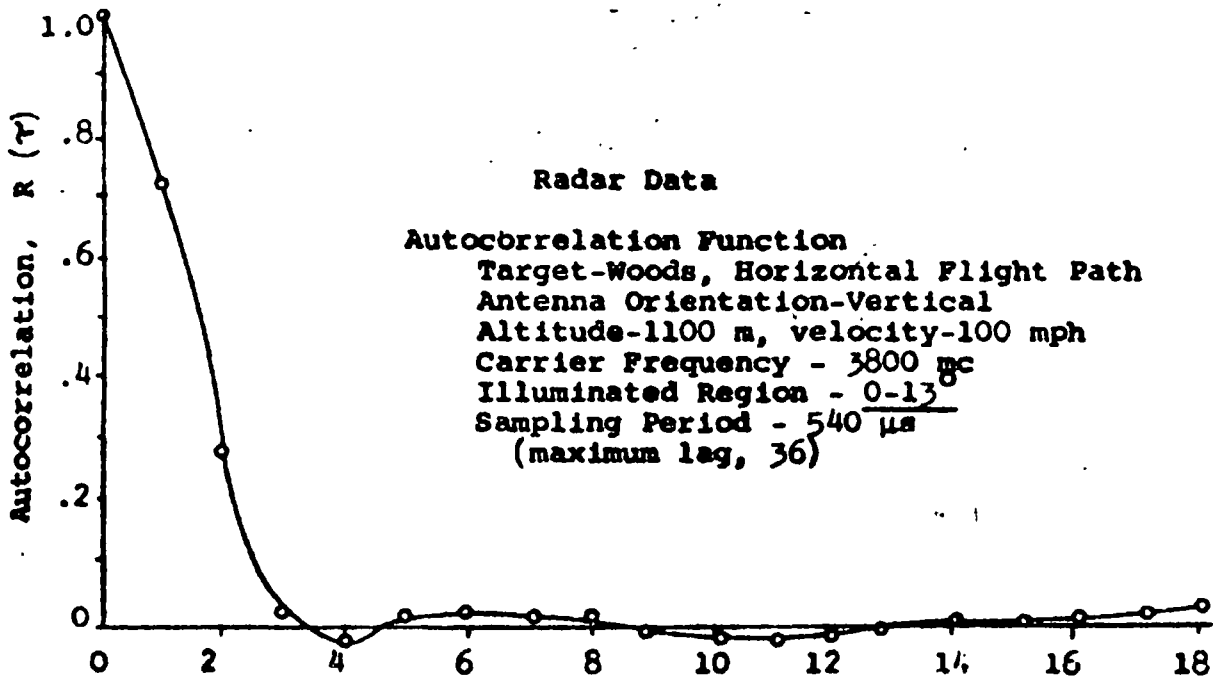




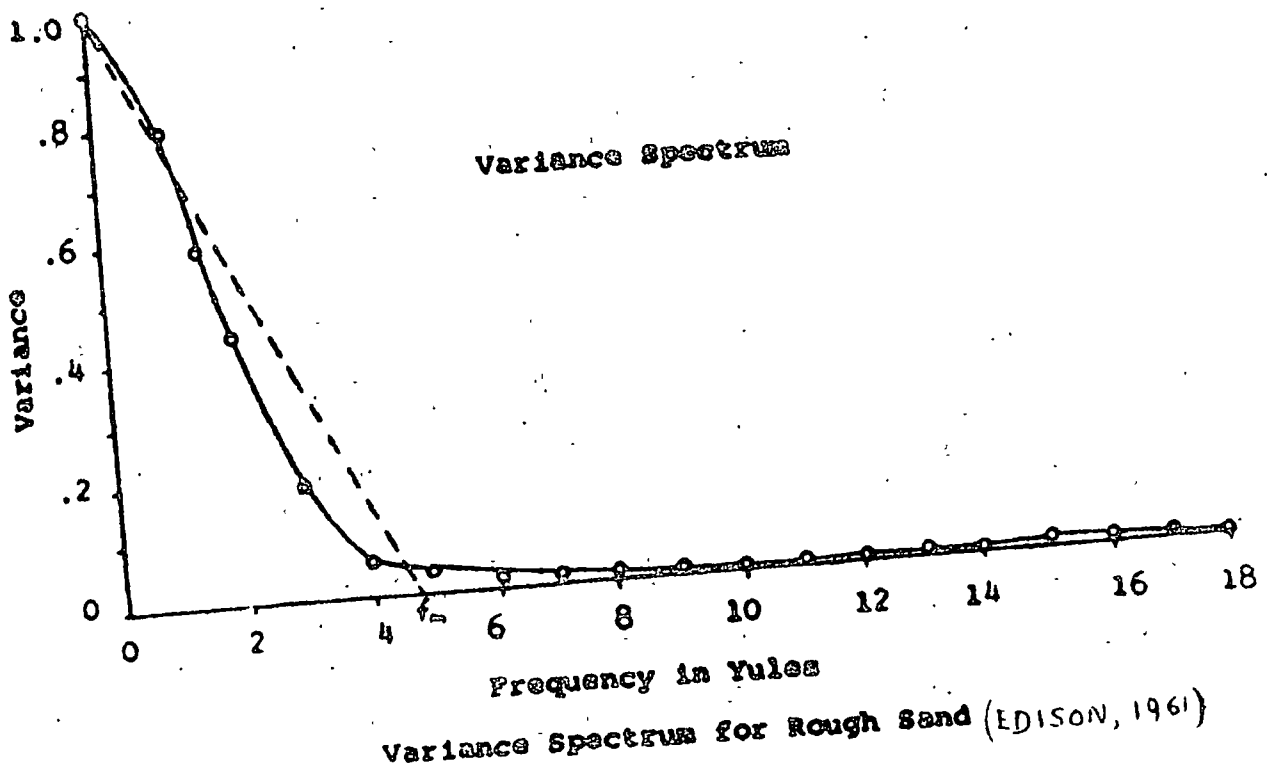
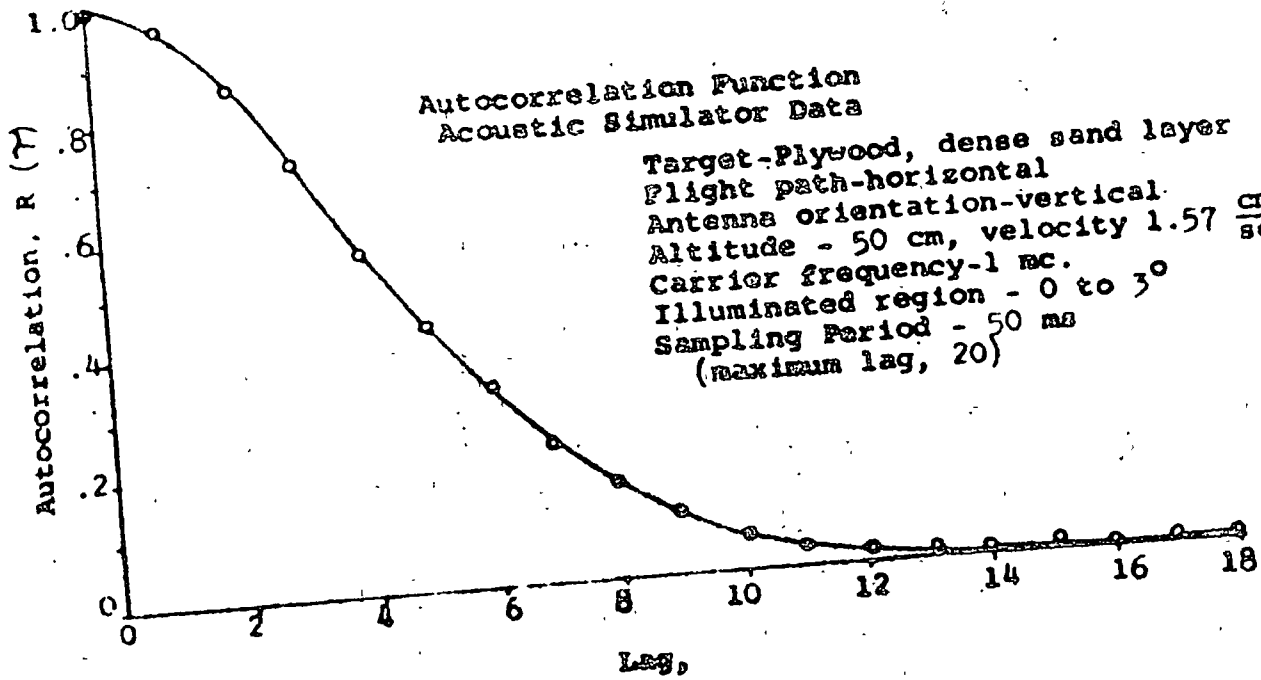
Reflection Coefficients for TM and TE Waves
Figure 3-8



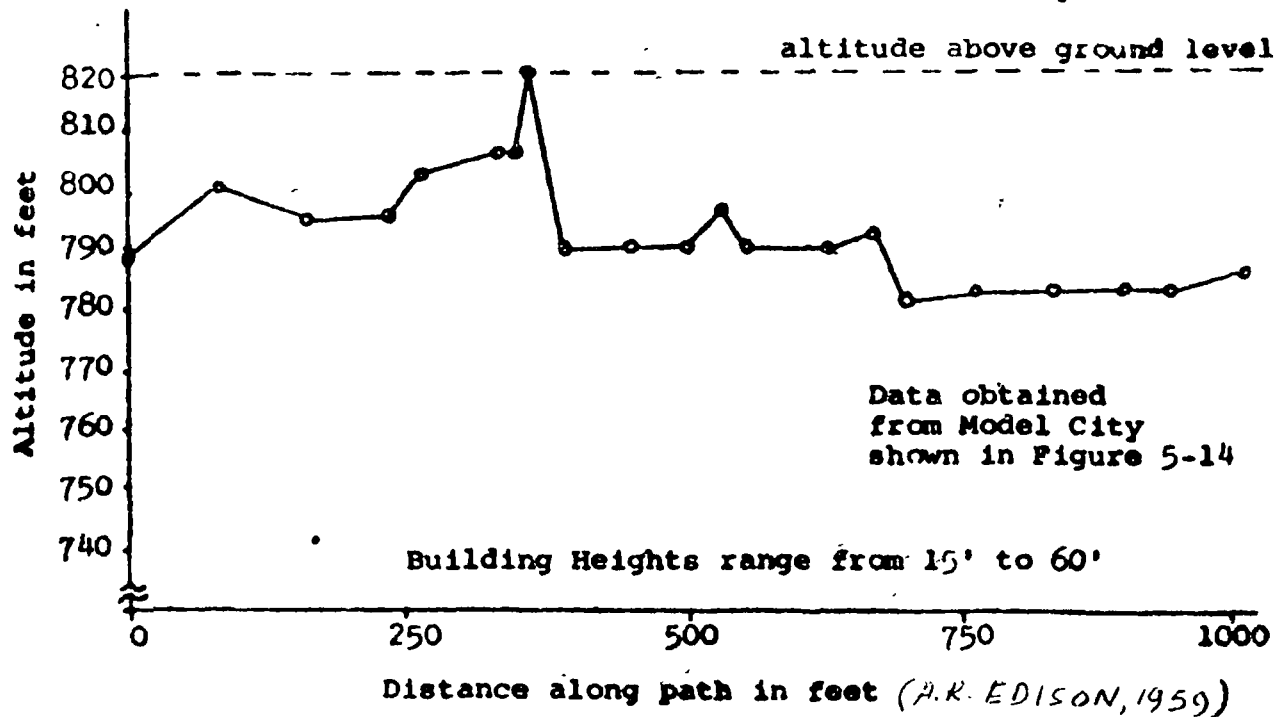
Reflection Coefficient for Acoustic Waves (EDISON, 1961)



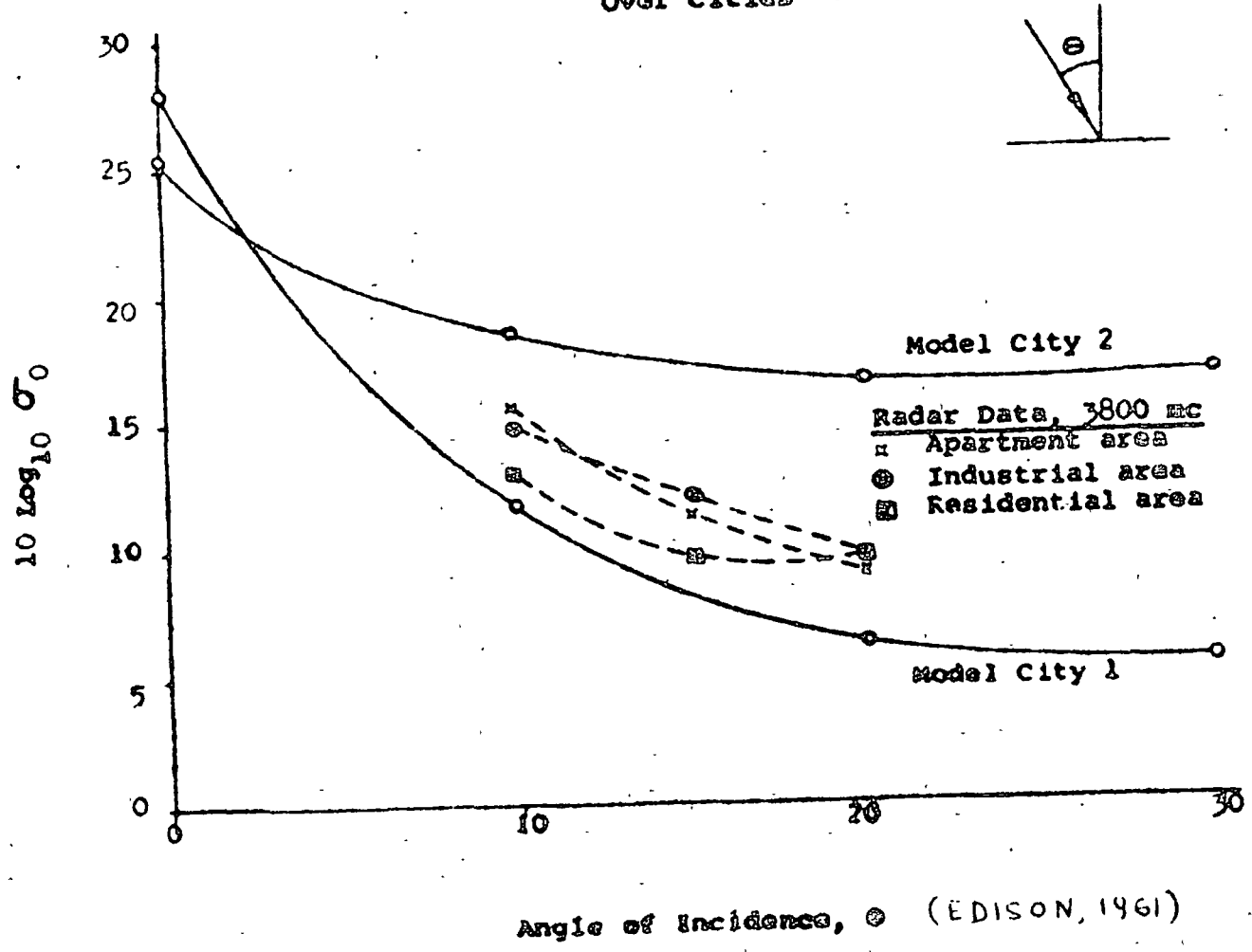
Variance Spectrum for Wooded Terrain (EDISON, 1961)



Pulse Radar Altitude Indication over City



Radar Backscattering Cross Sections Over Cities



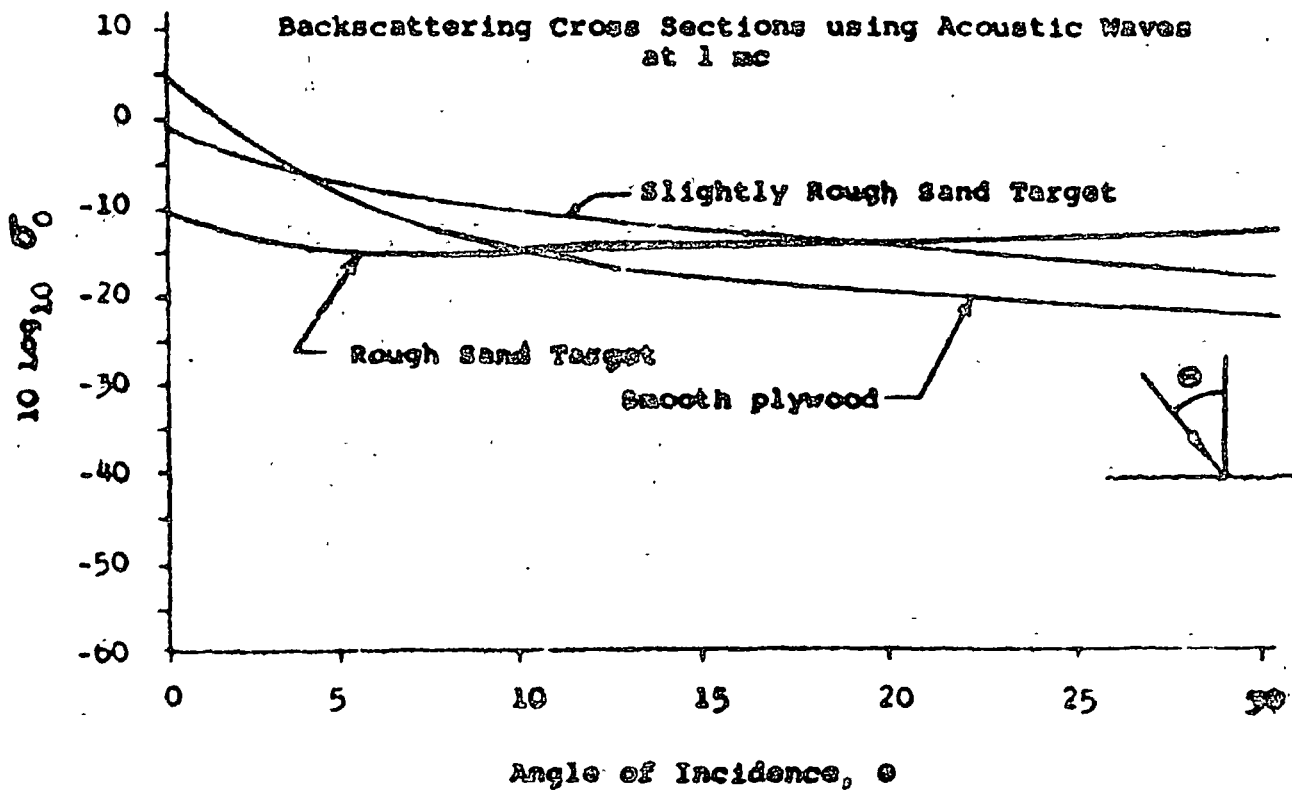
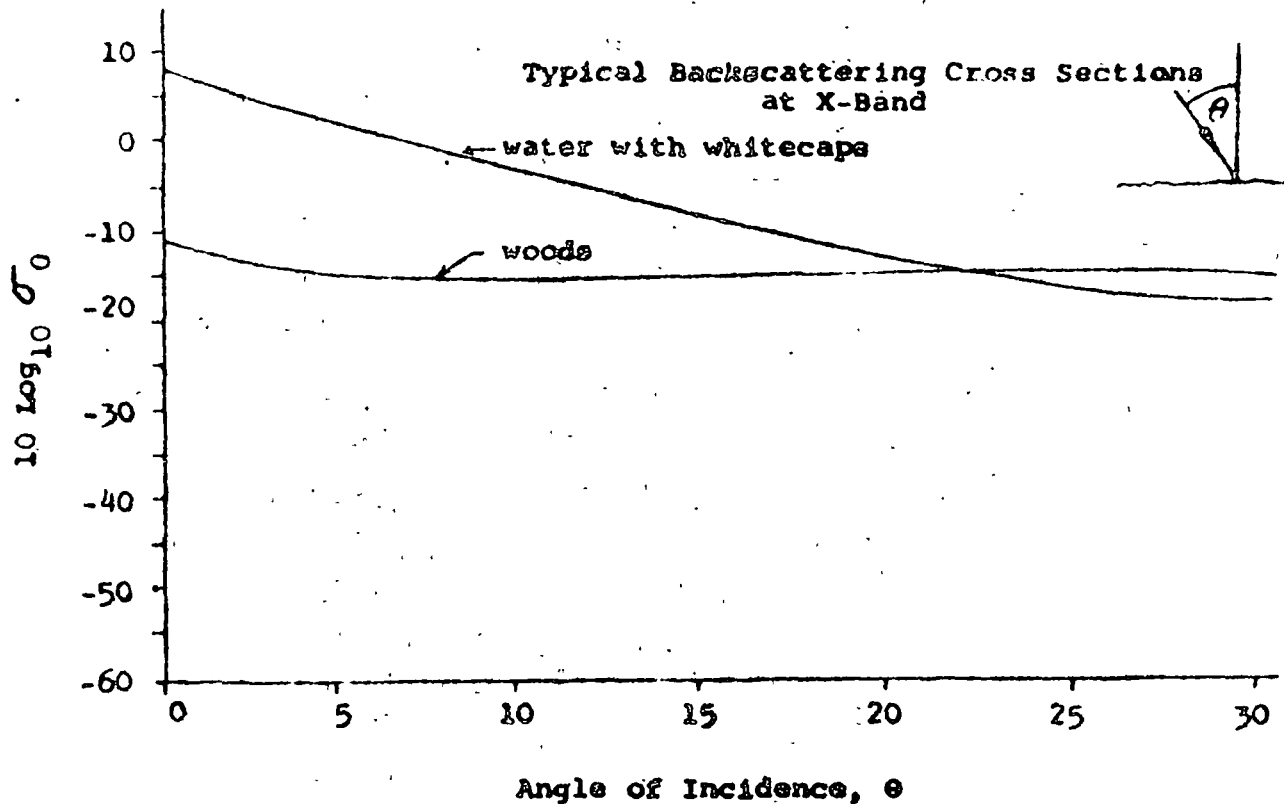
TARGET RECOMMENDATIONS (EDISON, 1961)

| Terrain | Target Base | Sand Particle size, wavelengths | Distance Between Particles, Wavelengths |
|----------------|----------------------------|---------------------------------|---|
| Woods | Plywood | 1-2 | 0 - 1/2 |
| Farmland | " | 1-2 | 1 - 5 |
| Desert | " | --- | ----- |
| Cities | " (buildings) ^a | $\frac{1}{2}$ -1 | 1 - 5 |
| Water (smooth) | galv. steel | $< \frac{1}{2}$ | 3 - 5 |
| Water (rough) | " " | $\frac{1}{2}$ -1 | 1 - 2 |
| Mountains | " " (shaped) ^b | $\frac{1}{2}$ -1 | 1 - 2 |

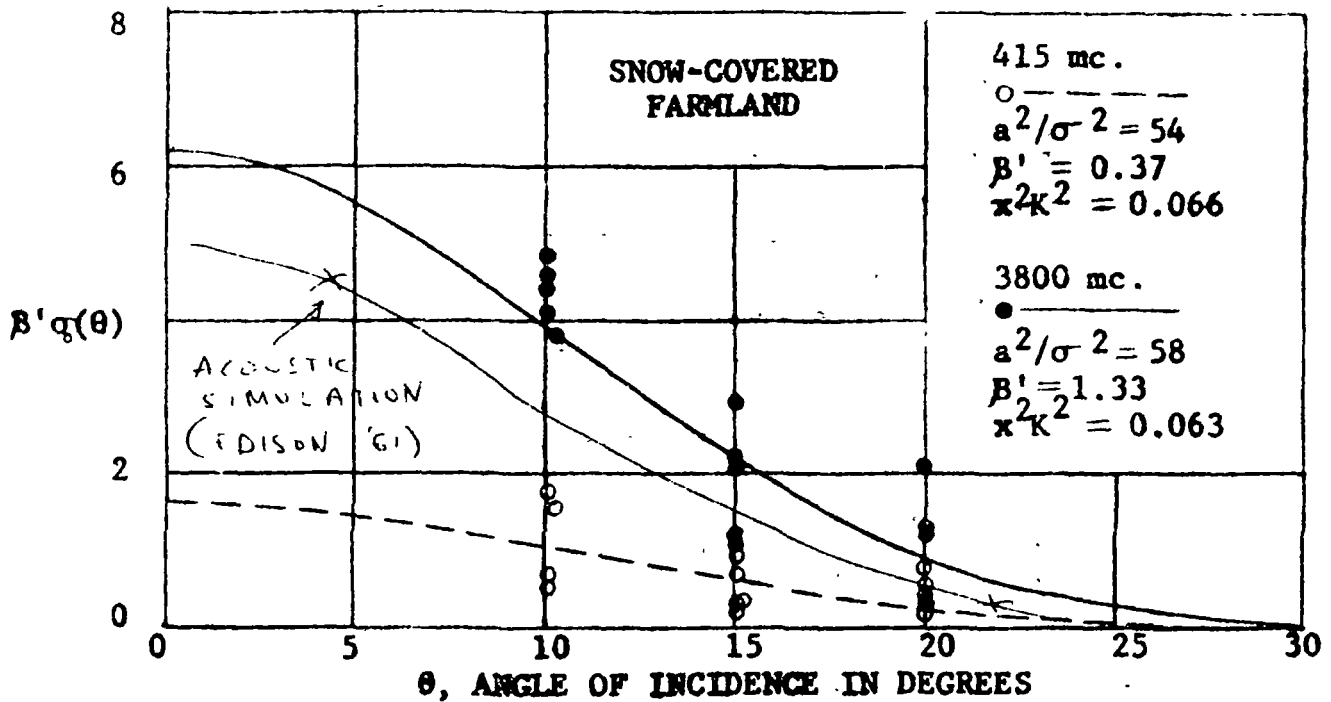
^aBuildings are pine blocks cut to size.

^bSteel can be formed into appropriate contours.

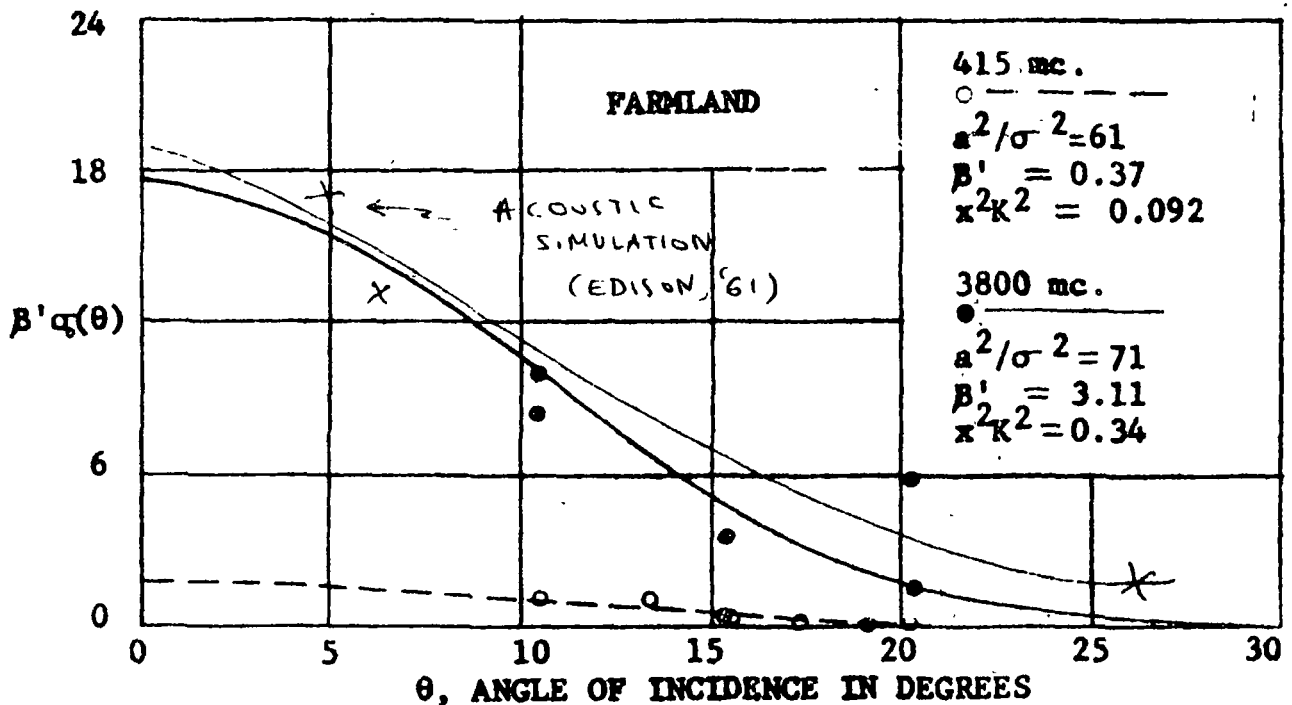
It should be observed that the slope of the radar backscattering cross section curves is important in modeling practice. The absolute level of the curves can be increased or decreased by proper scaling.



Typical Radar Back-Scattering Cross Sections (EDISON, 1961)

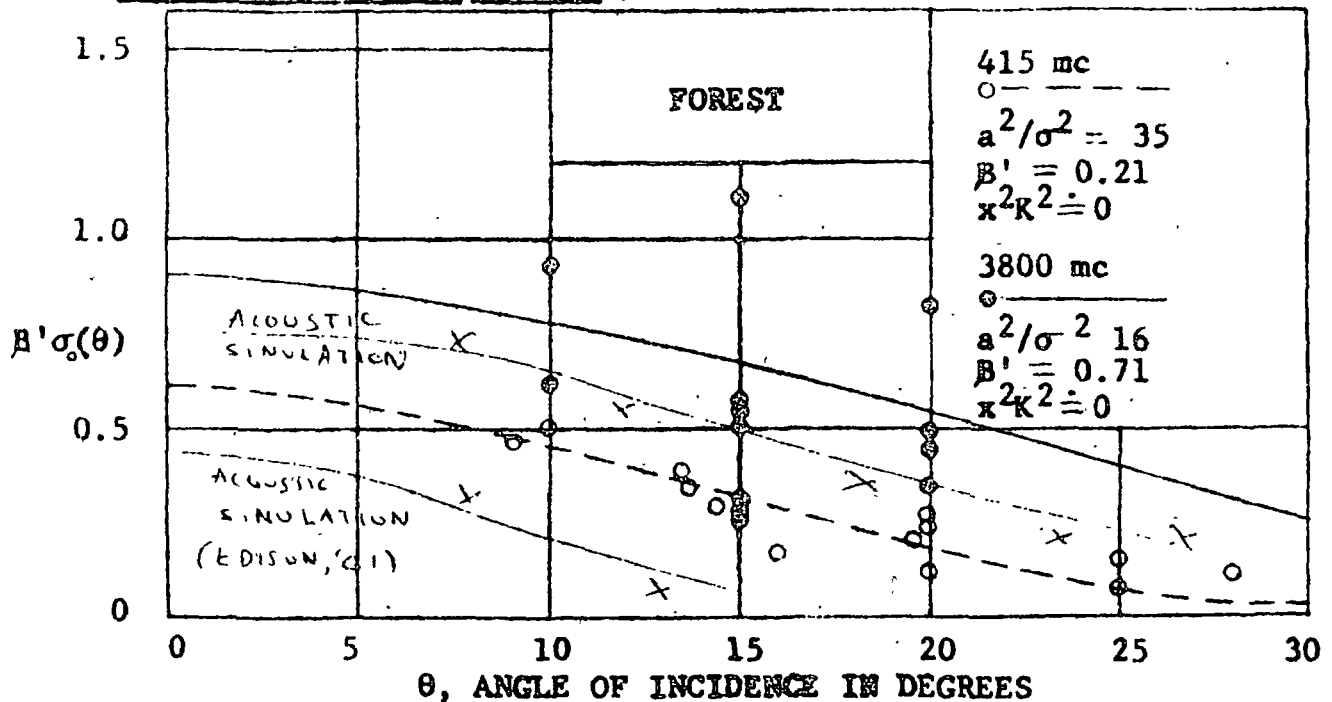


Median scattering curves for snow-covered farmland near Wahpeton, North Dakota. The target area was flat crop land containing two dry stream beds and was covered by eight inches of dry snow. (EDISON, '59).



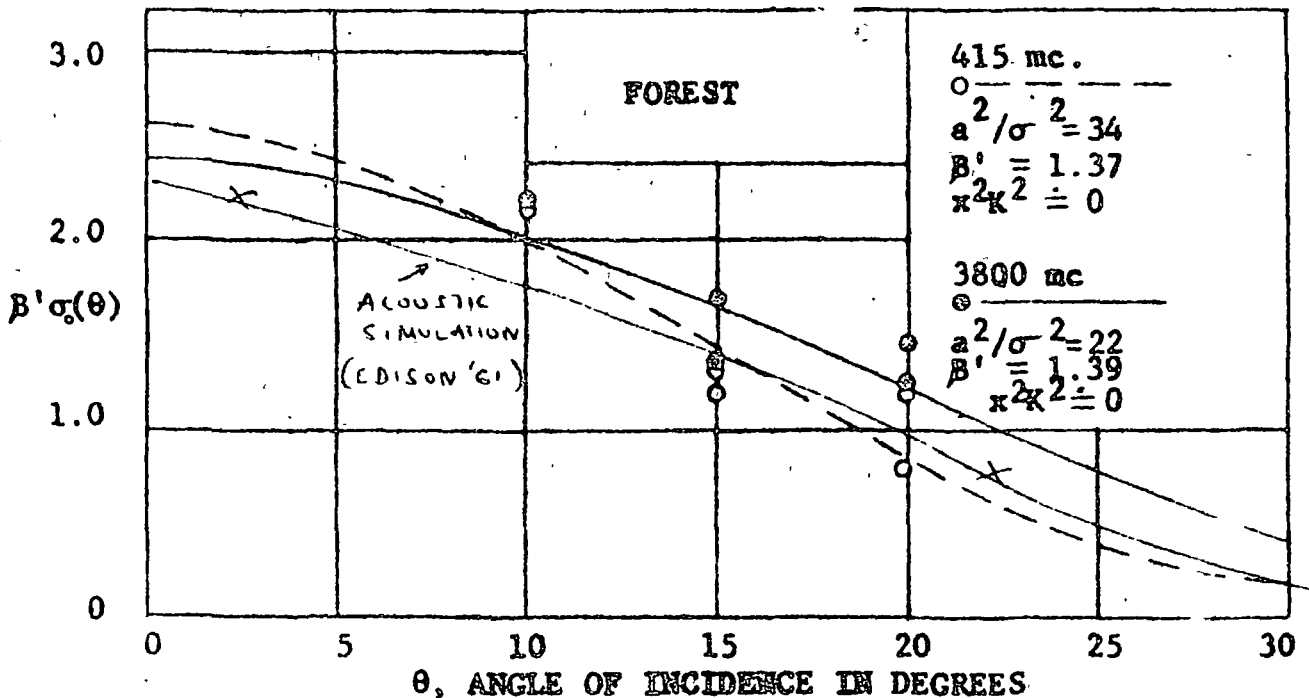
Median scattering curves for farmland near Cameron, Missouri. The target area was flat pasture and crop land containing a single line railroad. (EDISON, '59)

7.0a Scattering Cross Sections



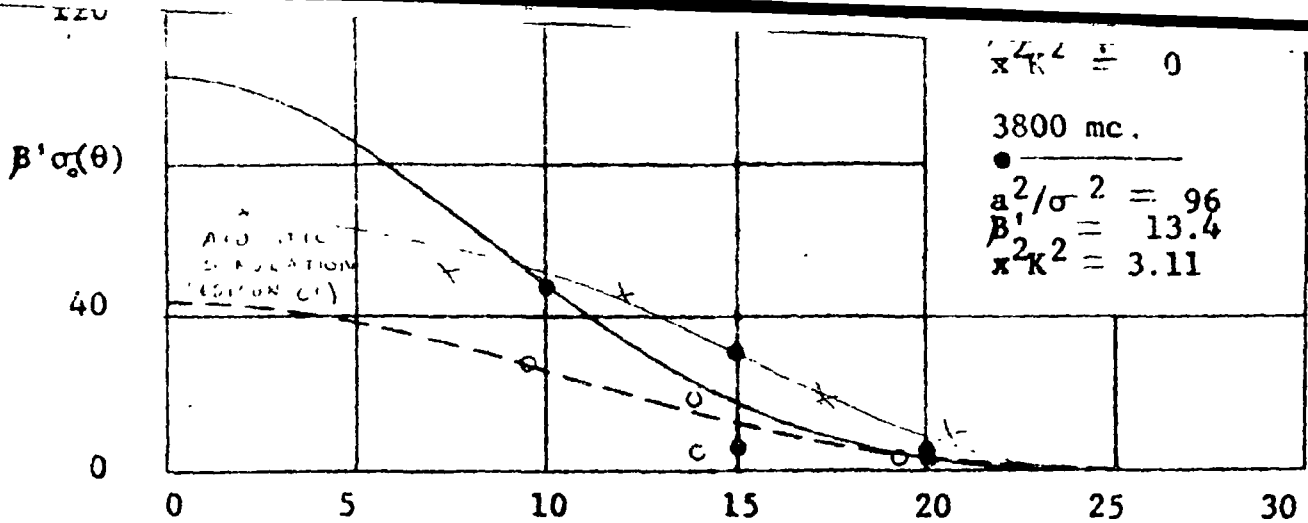
Median scattering curves for the forest at Pine Island, Minnesota. The target area was very flat and densely covered with pine, hemlock, birch, white ash, and elm trees from 20 to 40 feet in height.

(EDISON '59)



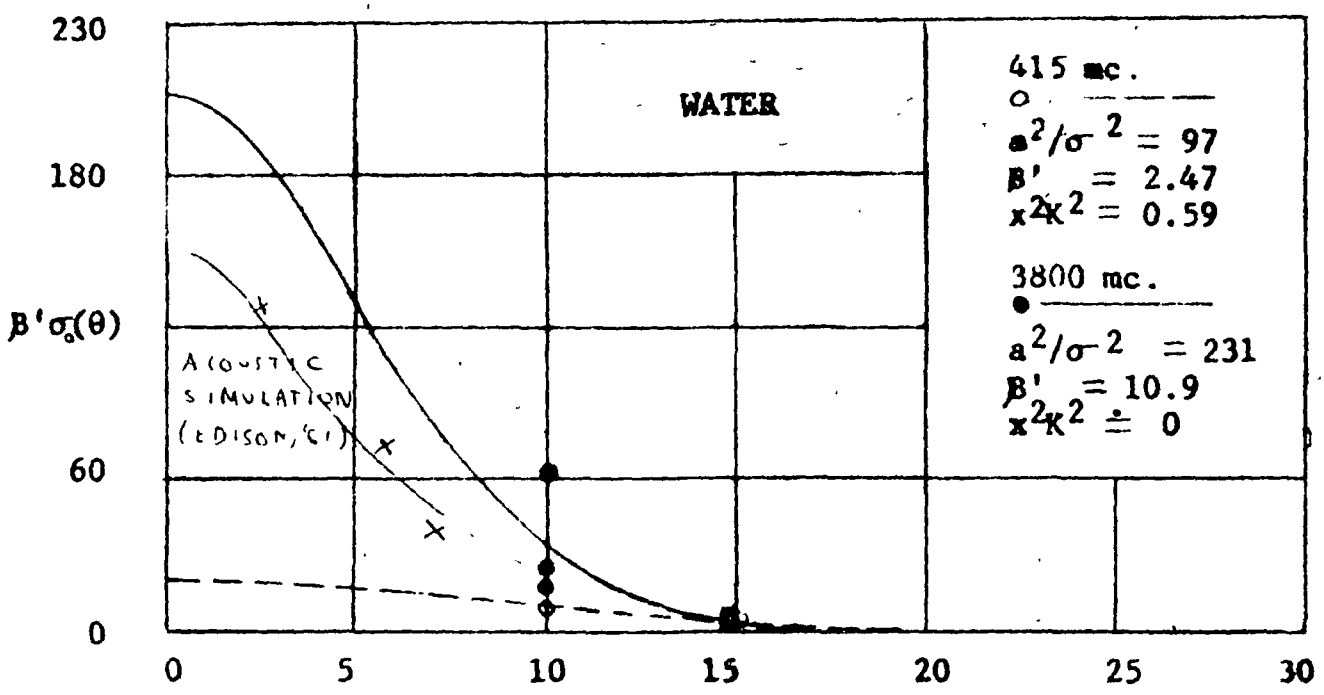
Median scattering curves for the forest at Presque Isle, Maine. The target had a snow-and-ice-covered rolling surface with a homogeneous covering of snow-bare ever-green fir and pine trees from 20 to 50 feet in height.

(EDISON '59)



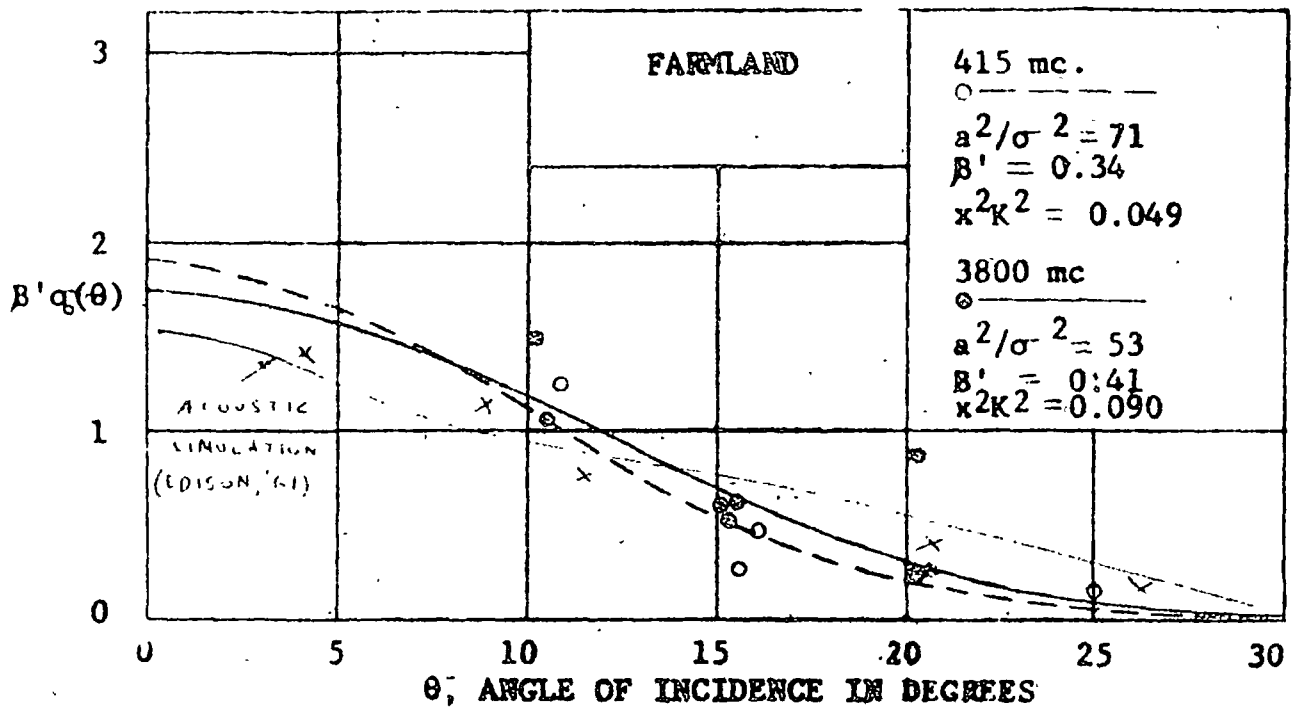
θ, ANGLE OF INCIDENCE IN DEGREES

Median scattering curves for a moderately rough surface on Lake Bemidji, Minnesota. The ripples and swells were about 15 to 20 inches vertically from peak to trough and three to four feet horizontally from peak to peak. (EDISON, '59)

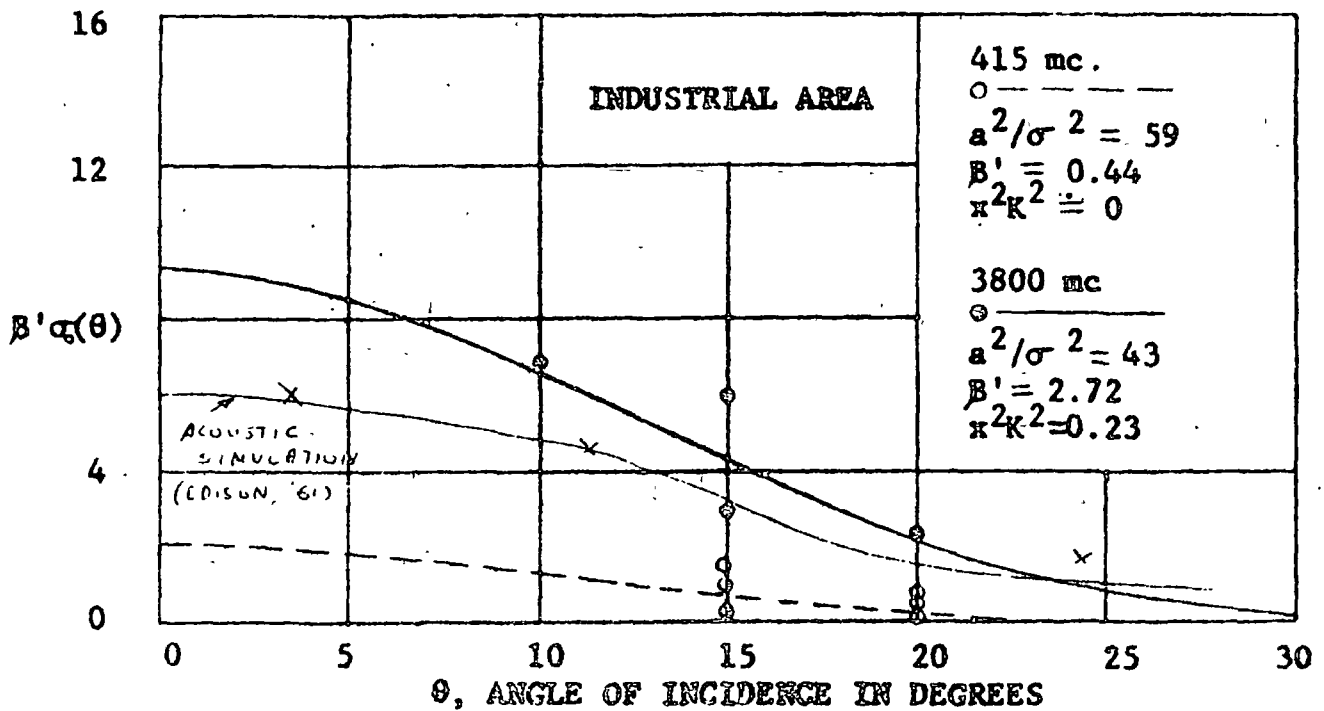


θ, ANGLE OF INCIDENCE IN DEGREES

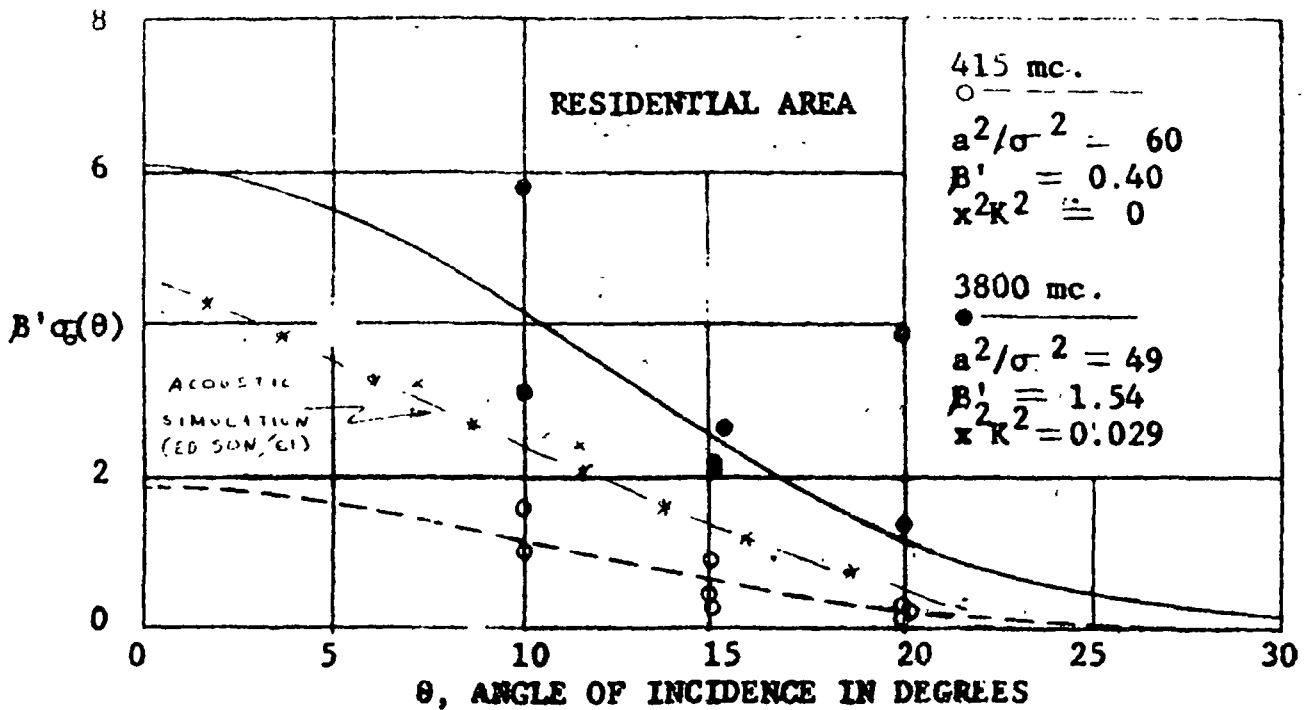
Median scattering curves for a relatively smooth surface on the Salton Sea in California. The air over the target was quite calm at the time of the flights. (EDISON, '59)



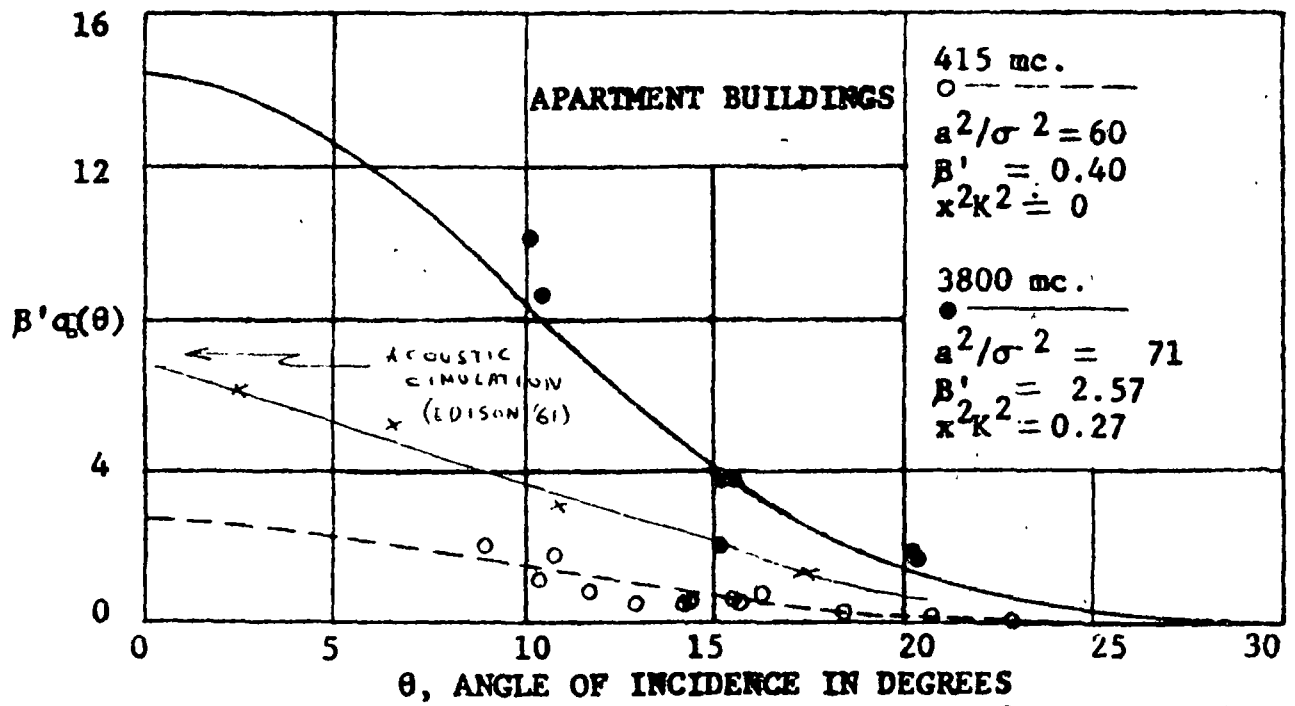
Median scattering curves for farmland near Sioux City, Iowa. The target area was flat crop land which had recently been plowed. (EDISON, '59)



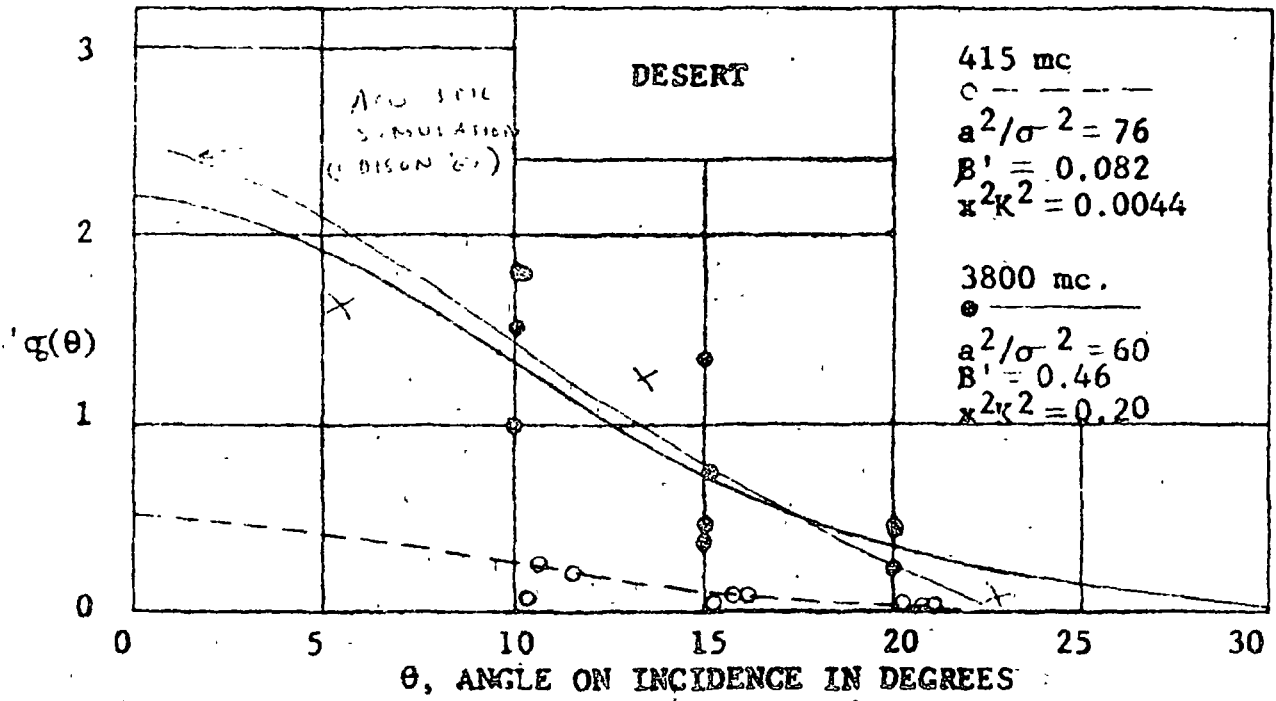
Median scattering curves for industrial area in Minneapolis, Minnesota. The target contained a predominant number of metal roofed factory buildings with a railroad yard at one edge. (EDISON, '59)



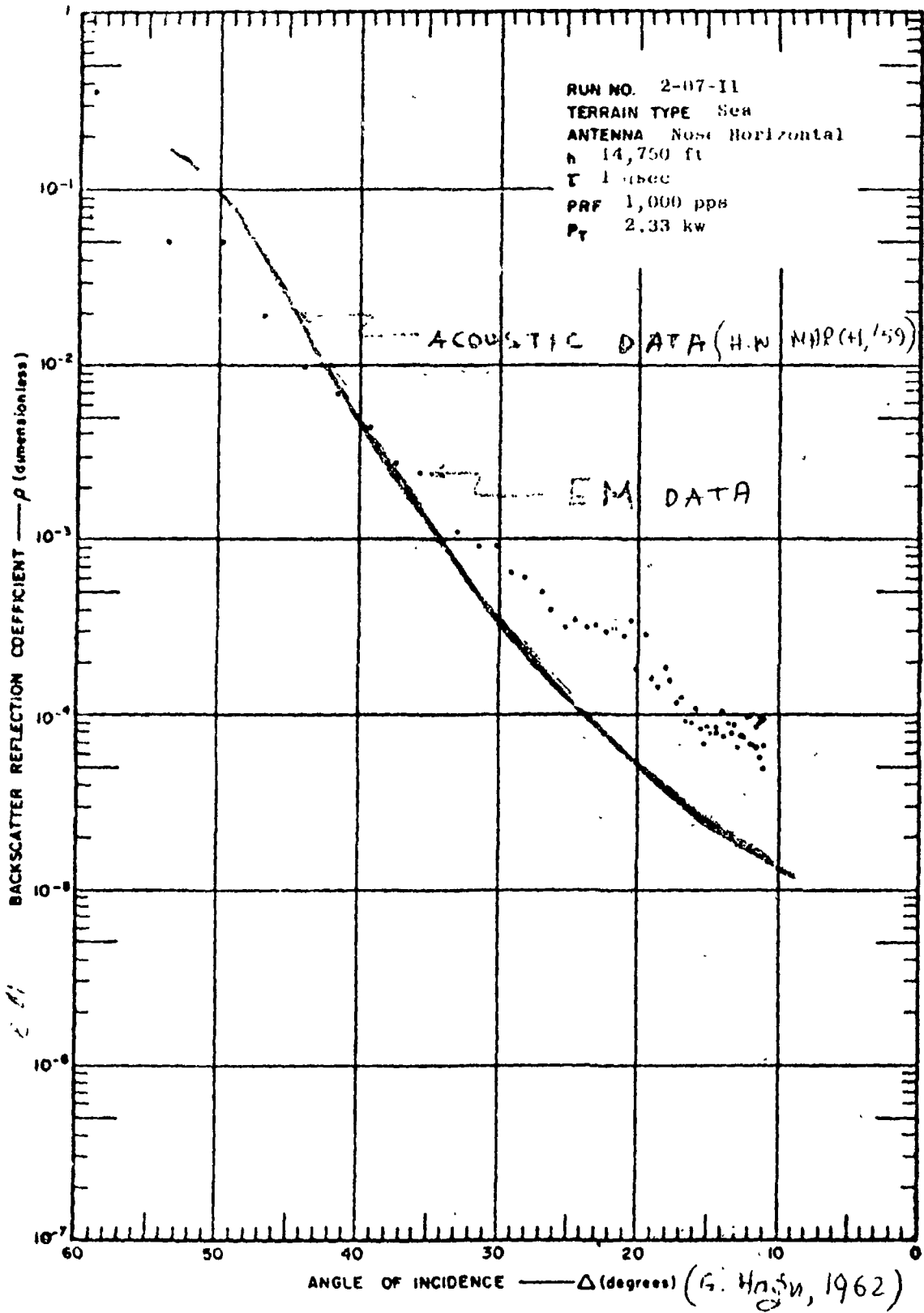
Median scattering curves for a residential area in Minneapolis, Minnesota. The target area was built up of one and two story brick and frame houses with pitched roofs, and had many old, well established trees. (EDISON, 1959)

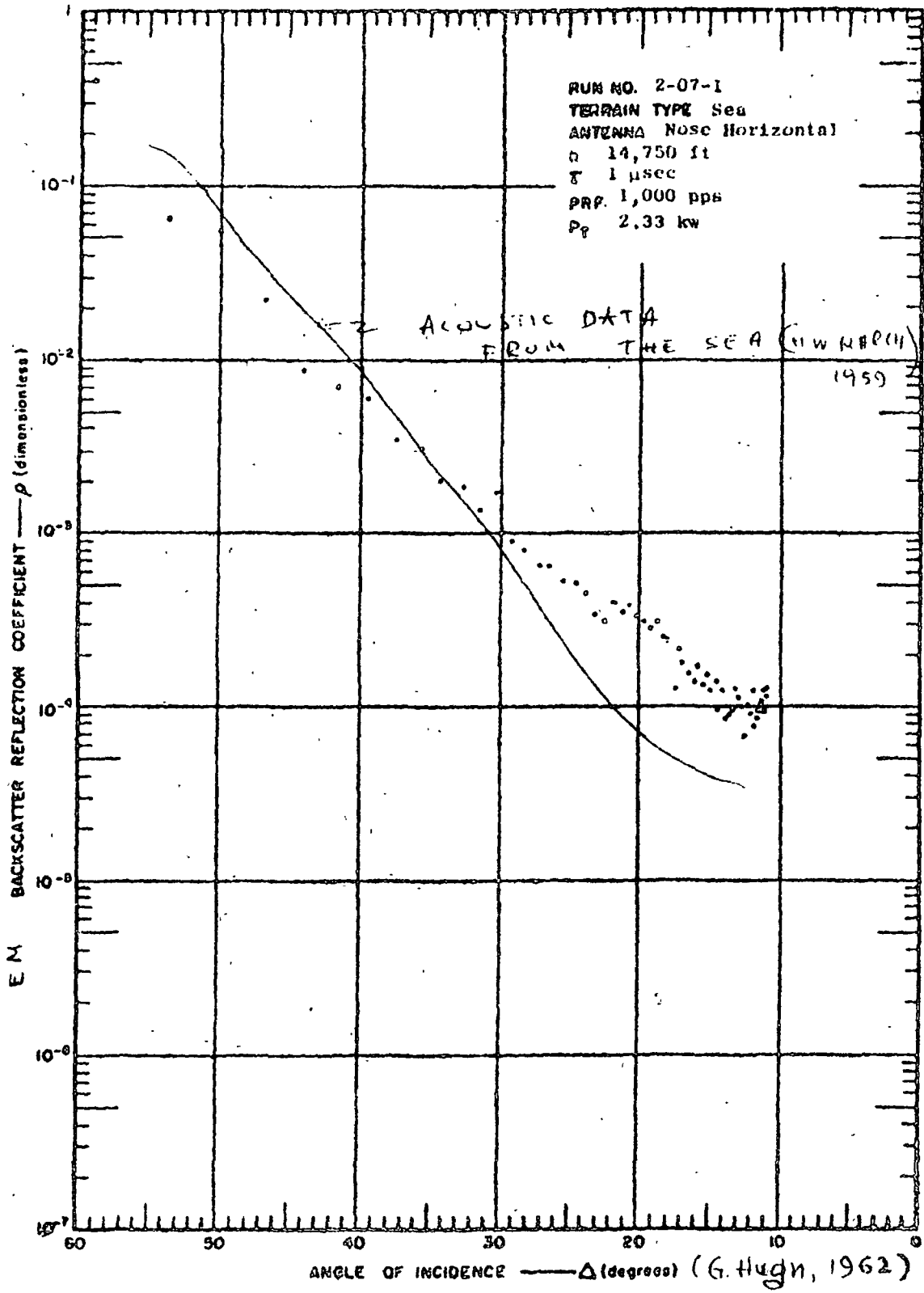


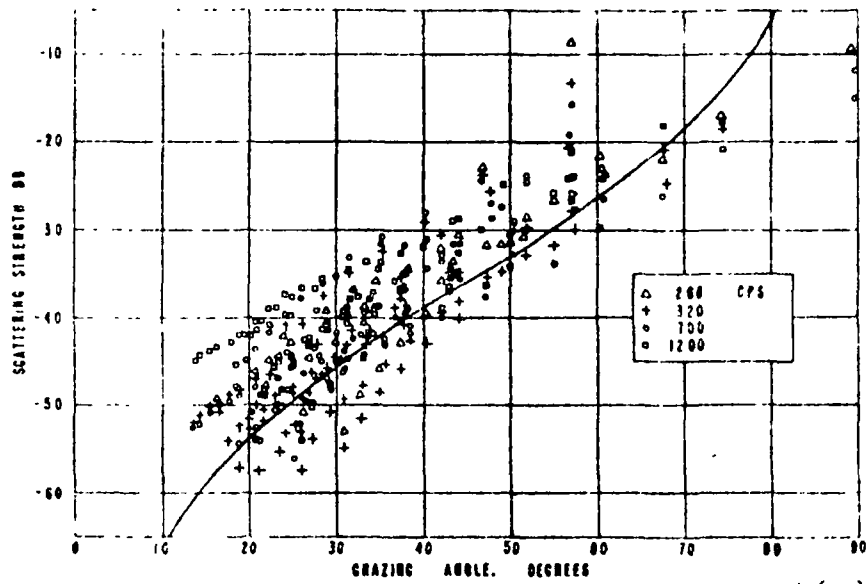
Median scattering curves for an area of apartment buildings in Kansas City, Missouri. The majority of the buildings were built of brick, flat roofed, and several stories tall. (EDISON, 1959)



Median scattering curves for a desert area near Salton Sea, California. The target area was flat, arid, sandy, and barren. (EDISON '59)

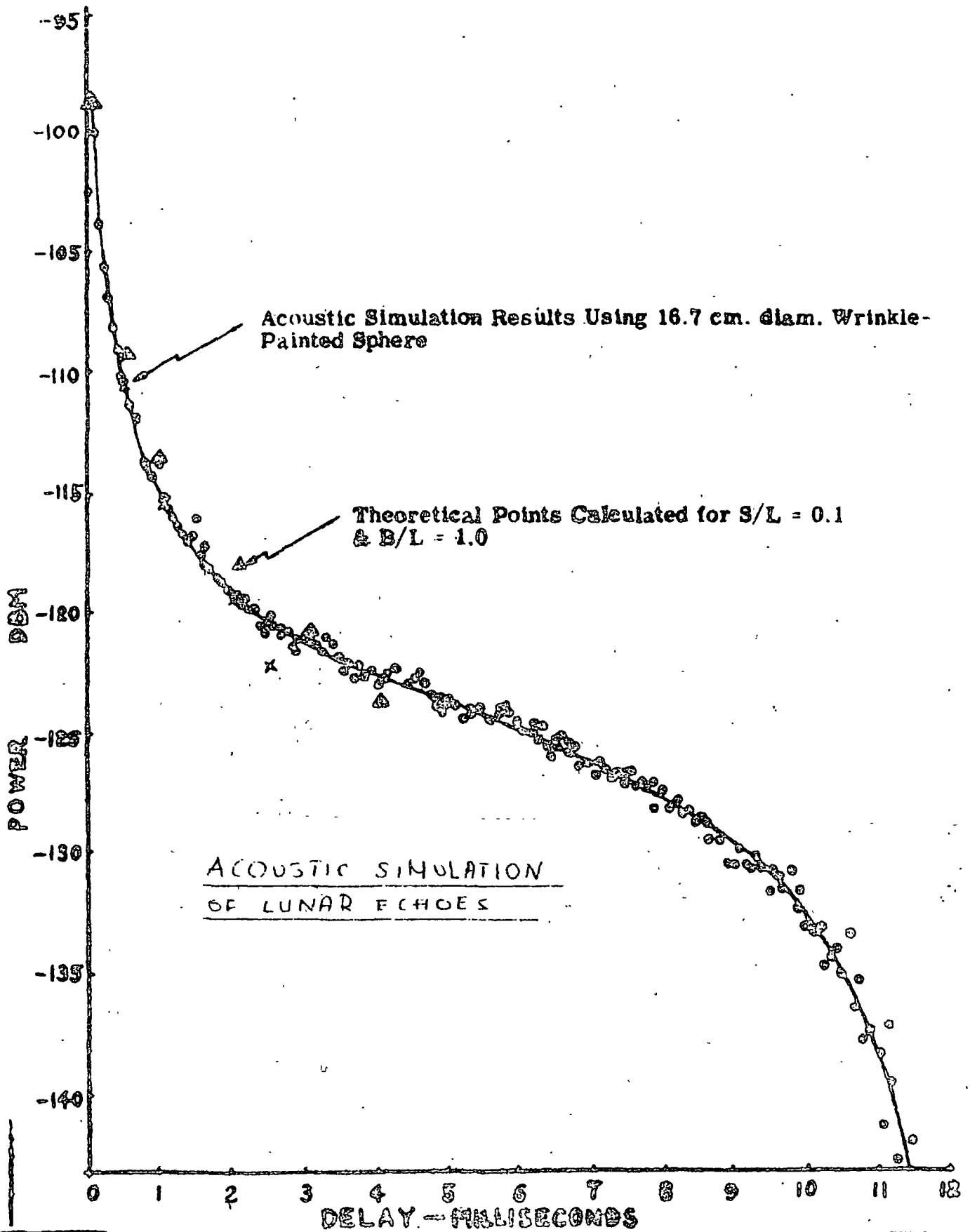






Scattering strength vs grazing angle. (H W March, '63)

ACOUSTIC SCATTERING FROM THE SEA



Mayre, H. S. "Acoustic Simulations of Lunar Echoes," Journal of Geophysical Research, Vol. 70, No. 16, (August, 1965), pp. 3831-3839.

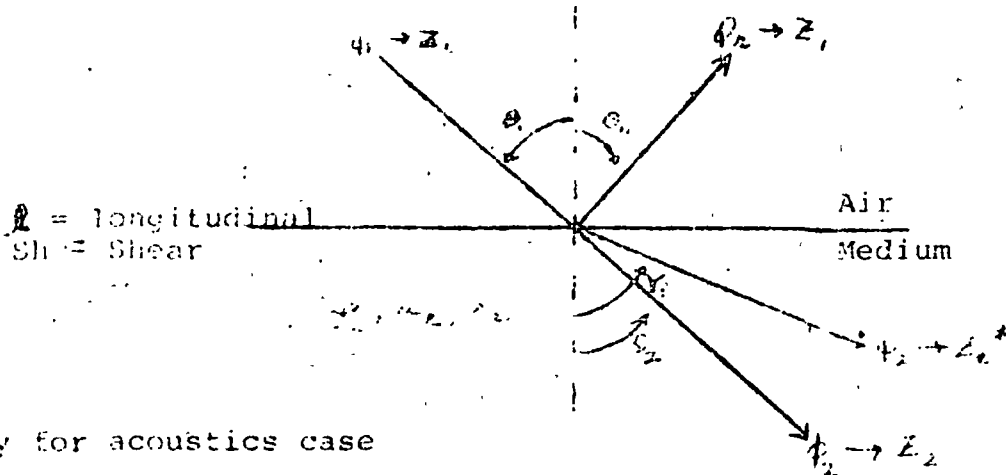
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COMPARISON OF REFLECTION TRANSMISSION COEFFICIENTS

| Acoustics (Liq/Solid) | | EMW (General) | |
|---|--|---|--|
| c^2 | $\frac{\lambda + 2\mu}{\rho}$ | | $\frac{1}{\epsilon \mu}$ |
| c | = Velocity of Propagation (longitudinal) | | |
| b^2 | = (shear) μ/ρ | | |
| | $\nabla^2 \phi + k^2 \phi = 0$ | | $\nabla^2 E + k^2 E = 0$ |
| | $\vec{V} = \nabla \phi + \nabla \times \psi$ | | $\vec{E} = -\nabla \phi + \frac{\partial \vec{A}}{\partial t}$ |
| <u>SNELL'S LAW</u> | | | |
| | $k_1 \sin \phi_1 = k_1 \sin \phi_{11} = k_2 \sin \gamma_2$ | | $k_1 \sin \phi = k_{11} \sin \phi_{11} = k_2 \sin \phi_2$ |
| Z_l | $Z_i = \rho c \sec \phi_1 \quad i = 1$ | $\sqrt{\frac{\mu}{\epsilon}} \sec \phi$ | Vertical Polarization |
| $Z_{sh} = Z_t$ | $\rho b \sec \gamma$ | $\sqrt{\frac{\mu}{\epsilon}} \cos \phi$ | Horizontal Polarization |
| Reflection Coefficient (longitudinal) | $V = \frac{Z_{tot} - Z_i}{Z_{tot} + Z_i}$ $i = 2$ | | $R = \frac{Z_2 - Z_1}{Z_2 + Z_1}$ |
| Transmission Coefficient (longitudinal) | $W = \frac{2(Z_1 \cos 2\gamma_2)}{Z_{tot} + Z_1}$ $i = 2$ | | $T = \frac{2Z_2}{Z_1 + Z_2}$ |
| Transmission Coefficient (shear) | $= \frac{2(Z_t \cos \gamma_2)}{Z_{tot} + Z_i} \quad i = 2$ | | |
| | $Z_{tot} = Z_1 \cos^2 2\gamma_2 + Z_t \sin^2 2\gamma_2$ | | |
| | $Z_i = \rho_i c_i / \cos \theta_i, \quad Z_t = \rho_2 b_2 / \cos \gamma_2$ | | |



*Only for acoustics case

Hook's Law

X_{ij} = stress in i to plane of application

j direction of applied force

strain
 $[X_{ij}] = [C_{ij}] [X_{ij}]$

$i = x_1 y_1 z$

$C_{ik} = C_{ki}$

$i = x_1 y_1 z$

$\Rightarrow 6 \text{ diag} + 15 \text{ off diag} = 21 \text{ constants}$

For isotropic case

$$[X_i] = [2a + \lambda, \lambda, \lambda] [X_i]$$

$$X_1 = (2\mu + \lambda) X_1 + \lambda X_2 + \lambda X_3 = 2a X_1 + \lambda (X_1 + X_2 + X_3)$$

μ, λ are Lamé constants

APPROXIMATE COUNT OF RCS MUS PEAKS
 (MILITARY AIRCRAFT SCOUT FACILITY PROJECT 1953)

| DB | ANGLE ⁰ | S. NO. | ANGLE ⁰ | DB | S. NO. | ANGLE ⁰ | DB | S. NO. | ANGLE ⁰ | DB | S. NO. | ANGLE ⁰ | |
|----|--------------------|--------|--------------------|------|--------|--------------------|------|--------|--------------------|------|--------|--------------------|------|
| 1 | 28.3 | 24 | 83 | 16 | 44 | 176 | 31.8 | 46 | 269 | 18.0 | 27 | 354 | 12.5 |
| 2 | 27.7 | 15 | 85 | 13.0 | 45 | 177 | 33.0 | 47 | 268 | 18.9 | 28 | 352.5 | 13.5 |
| 3 | 26.7 | 16 | 88 | 12.5 | 46 | 180 | 41.1 | 48 | 270 | 34.0 | 29 | 352 | 14.4 |
| 4 | 26.1 | 17 | 89 | 17.0 | | | | | | | 30 | 354 | 15.0 |
| 5 | 24.6 | 18 | 90 | 36.0 | 0 | 180 | 41.1 | 0 | 270 | 34.0 | 31 | 355 | 18.0 |
| 6 | 24.6 | 0 | 90 | 36.0 | 1 | 182 | 33.5 | 1 | 271 | 24.6 | 32 | 357 | 23.0 |
| 7 | 24.8 | 1 | 94 | 17.6 | 2 | 184 | 28.2 | 2 | 272 | 22.6 | 33 | 360 | 28.0 |
| 8 | 17.3 | 2 | 95 | 13.5 | 3 | 186 | 31.3 | 3 | 274 | 18.6 | | | |
| 9 | 17.2 | 3 | 96 | 15.6 | 4 | 187 | 32.2 | 4 | 275 | 13.0 | | | |
| 10 | 18.5 | 4 | 99 | 17.5 | 5 | 189 | 31.5 | 5 | 277 | 13.0 | | | |
| 11 | 19 | 5 | 101 | 20.6 | 6 | 190 | 27.9 | 6 | 280 | 15.0 | | | |
| 12 | 20 | 6 | 102 | 21.4 | 7 | 192 | 27.5 | 7 | 282 | 19.0 | | | |
| 13 | 22 | 7 | 107 | 17.0 | 8 | 193 | 31.5 | 8 | 285 | 17.0 | | | |
| 14 | 24.5 | 8 | 108 | 18.7 | 9 | 194 | 30.5 | 9 | 286 | 18.0 | | | |
| 15 | 26.2 | 9 | 110 | 19.5 | 10 | 195 | 27.0 | 10 | 288 | 14.6 | | | |
| 16 | 28 | 10 | 112 | 16.5 | 11 | 197 | 28.1 | 11 | 290 | 15.6 | | | |
| 17 | 31 | 11 | 114 | 16.5 | 12 | 199 | 30.5 | 12 | 293 | 15.0 | | | |
| 18 | 33 | 12 | 116 | 13.6 | 13 | 201 | 23.5 | 13 | 294 | 16.4 | | | |
| 19 | 33.8 | 13 | 118 | 14.4 | 14 | 203 | 26.8 | 14 | 296 | 14.5 | | | |
| 20 | 35 | 14 | 120 | 17.6 | 15 | 207 | 21.0 | 15 | 298 | 12.0 | | | |
| 21 | 35 | 15 | 121 | 16.0 | 16 | 209 | 26.5 | 16 | 300 | 13.5 | | | |
| 22 | 37 | 16 | 124 | 21.8 | 17 | 210 | 23.0 | 17 | 304 | 31.4 | | | |
| 23 | 38 | 17 | 126 | 20.3 | 18 | 212 | 21.0 | | | | | | |
| 24 | 41 | 18 | 127 | 18.5 | 19 | 213 | 23.3 | 0 | 304 | 31.4 | | | |
| 25 | 43 | 19 | 128 | 20.5 | 20 | 216 | 24.0 | 1 | 305 | 19.6 | | | |
| 26 | 44 | 20 | 132 | 22.0 | 21 | 218 | 20.7 | 2 | 307 | 9.5 | | | |
| 27 | 45.6 | 21 | 134 | 19.7 | 22 | 222 | 23.3 | 3 | 308 | 9.5 | | | |
| 28 | 46 | 22 | 135 | 18.5 | 23 | 223 | 23.2 | 4 | 310 | 9.0 | | | |
| 29 | 48 | 23 | 136 | 15.6 | 24 | 225 | 20.0 | 5 | 311 | 9.4 | | | |
| | | | | | 25 | 228 | 22.0 | 6 | 313 | 13.5 | | | |
| 30 | 49 | 24 | 138 | 16.3 | | | | | | | | | |
| 31 | 51 | 25 | 140 | 21.0 | 26 | 230 | 24.7 | 7 | 314 | 10.6 | | | |
| 32 | 52 | 26 | 142 | 19.5 | 27 | 232 | 23.5 | 8 | 315 | 11 | | | |
| 33 | 54 | 27 | 144 | 19.4 | 28 | 233 | 20.3 | 9 | 317 | 11 | | | |
| 34 | 56 | 28 | 145 | 21.5 | 29 | 235 | 17.3 | 10 | 319 | 8.8 | | | |
| | | 29 | 146 | 22.5 | 30 | 237 | 21.9 | 11 | 320 | 6.5 | | | |
| 0 | 56 | 30 | 148 | 19.6 | 31 | 238 | 21.5 | 12 | 322 | 13.5 | | | |
| 1 | 58 | 31 | 150 | 23.2 | 32 | 240 | 17.5 | 13 | 324 | 9.5 | | | |
| 2 | 62 | 32 | 152 | 23.6 | 33 | 242 | 16.7 | 14 | 326 | 7.5 | | | |
| 3 | 63 | 33 | 154 | 16.5 | 34 | 244 | 19.6 | 15 | 327 | 6.5 | | | |
| | | | | | 35 | 245 | 22.8 | 16 | 329 | 9.5 | | | |
| 4 | 65 | 34 | 158 | 25.3 | | | | | | | | | |
| 5 | 66 | 35 | 160 | 25.3 | 36 | 248 | 22.5 | 17 | 331 | 9.0 | | | |
| 6 | 68 | 36 | 161 | 23.7 | 37 | 250 | 24.0 | 18 | 332 | 5.5 | | | |
| 7 | 70 | 37 | 163 | 27.0 | 38 | 251 | 23.0 | 19 | 334 | 6.5 | | | |
| 8 | 72 | 38 | 166 | 27.0 | 39 | 252 | 23.0 | 20 | 335 | 8.8 | | | |
| 9 | 73 | 39 | 167 | 28.3 | 40 | 254 | 21.2 | 21 | 337 | 13.3 | | | |
| | | 40 | 168 | 27.5 | 41 | 257 | 19.8 | 22 | 340 | 14.4 | | | |
| 10 | 74 | 41 | 170 | 28.3 | 42 | 258 | 18.5 | 23 | 342 | 17.3 | | | |
| 11 | 76.8 | 42 | 172 | 31.5 | 43 | 261 | 16.5 | 24 | 344 | 17.8 | | | |
| 12 | 80 | 43 | 174 | 25.7 | 44 | 263 | 13.3 | 25 | 348 | 13.4 | | | |
| 13 | 81 | | | | 45 | 265 | 12.5 | 26 | 349 | 12.5 | | | |

or 6

APPROXIMATE COUNT OF RCS PEAKS
(AF-MDC-HDRP-RAT SCMT FACILITY PROJECT 6503)

| S. NO. | ANGLE ⁰ | DB | S. NO. | ANGLE ⁰ | DB | S. NO. | ANGLE ⁰ | DB | S. NO. | ANGLE ⁰ | DB |
|--------|--------------------|------|--------|--------------------|------|--------|--------------------|------|--------|--------------------|------|
| 0 | | 27.2 | 50 | 54 | 13.2 | 0 | 90 | 34 | 50 | 125.9 | 21.3 |
| 1 | 2 | 23.6 | 51 | 55 | 20.7 | 1 | 90.5 | 23.6 | 51 | 126 | 21.5 |
| 2 | 3.7 | 18 | 52 | 56 | 28.1 | 2 | 91 | 20.3 | 52 | 126.5 | 19.7 |
| 3 | 5 | 25.5 | | | | 3 | 92 | 21 | 53 | 127.9 | 19.7 |
| 4 | 6 | 18.7 | 0 | 56 | 28.1 | 4 | 92.5 | 20.5 | 54 | 128 | 21 |
| 5 | 7.8 | 10.2 | 1 | 56.7 | 21.2 | 5 | 93 | 18.5 | 55 | 128.5 | 20.3 |
| 6 | 10 | 12.4 | 2 | 58 | 16.1 | 6 | 94 | 17.7 | 56 | 129.9 | 18.6 |
| 7 | 11 | 10.6 | 3 | 58.3 | 14.5 | 7 | 94.5 | 17 | 57 | 131 | 16.3 |
| 8 | 11.9 | 12.3 | 4 | 59 | 12.7 | 8 | 95 | 14.6 | 58 | 131.5 | 17.2 |
| 9 | 13 | 16.1 | 5 | 59.9 | 11.8 | 9 | 96.5 | 15 | 59 | 132.5 | 18.7 |
| 10 | 14.2 | 15.7 | 6 | 61.5 | 9.1 | 10 | 97 | 12.8 | 60 | 132.8 | 18.4 |
| 11 | 16 | 17.5 | 7 | 63 | 15.2 | 11 | 98.3 | 11.9 | 61 | 134 | 19.7 |
| 12 | 16.2 | 18 | 8 | 64.2 | 18.1 | 12 | 99 | 12.5 | 62 | 136.7 | 18.6 |
| 13 | 17 | 17.7 | 9 | 65 | 13.8 | 13 | 100 | 14.5 | 63 | 137.9 | 10.5 |
| 14 | 17.6 | 17.4 | 10 | 65.5 | 13.7 | 14 | 100.9 | 18.8 | 64 | 138.3 | 14.2 |
| 15 | 19 | 15.6 | 11 | 65.9 | 14.7 | 15 | 101 | 18.6 | 65 | 139 | 10.3 |
| 16 | 20.1 | 16.8 | 12 | 66 | 16 | 16 | 102 | 17.5 | 66 | 140.9 | 19.0 |
| 17 | 22 | 13.6 | 13 | 67 | 11 | 17 | 102.3 | 17 | 67 | 141 | 21.4 |
| 18 | 22.2 | 10.3 | 14 | 68 | 14.2 | 18 | 103.8 | 17.7 | 68 | 142.5 | 21.2 |
| 19 | 24.1 | 9.8 | 15 | 68.8 | 14 | 19 | 104.1 | 18 | 69 | 143 | 22.1 |
| 20 | 25.6 | 10.7 | 16 | 69 | 13 | 20 | 105 | 21.6 | 70 | 144 | 18.2 |
| 21 | 26 | 11.7 | 17 | 70.5 | 15.1 | 21 | 106 | 21.5 | 71 | 144.5 | 17.5 |
| 22 | 27 | 8.2 | 18 | 71 | 16.4 | 22 | 106.5 | 22.9 | 72 | 145 | 20.6 |
| 23 | 28 | 11.8 | 19 | 72 | 15.8 | 23 | 107 | 21.5 | 73 | 145.5 | 15.8 |
| 24 | 29 | 7.8 | 20 | 72.8 | 17.2 | 24 | 107.5 | 20.3 | 74 | 146.5 | 18.1 |
| 25 | 31 | 11.3 | 21 | 73 | 17.2 | 25 | 108 | 20.0 | 75 | 147 | 19.2 |
| 26 | 32.2 | 7.7 | 22 | 73.5 | 17.2 | 26 | 109 | 20.3 | 76 | 148.5 | 21.3 |
| 27 | 33 | 11.8 | 23 | 73.9 | 17.2 | 27 | 109.5 | 18.8 | 77 | 149 | 21.9 |
| 28 | 33.5 | 11.8 | 24 | 74.1 | 18.2 | 28 | 111 | 14.1 | 78 | 149.5 | 21 |
| 29 | 34.1 | 10.2 | 25 | 75 | 15.7 | 29 | 112.2 | 10.6 | 79 | 150.5 | 16.8 |
| 30 | 36 | 8.8 | 26 | 76 | 18.4 | 30 | 113 | 13.7 | 80 | 151 | 16 |
| 31 | 36.5 | 7.7 | 27 | 77 | 19.5 | 31 | 113.9 | 14.9 | 81 | 151.9 | 13.6 |
| 32 | 37 | 8.5 | 28 | 77.5 | 21.6 | 32 | 114.2 | 15.7 | 82 | 152.5 | 21.2 |
| 33 | 38.2 | 7.7 | 29 | 78.9 | 19.3 | 33 | 115 | 11 | 83 | 153.8 | 22.8 |
| 34 | 39.8 | 7 | 30 | 79 | 18.7 | 34 | 116 | 15.5 | 84 | 155.5 | 21.1 |
| 35 | 40.2 | 7.5 | 31 | 80 | 18 | 35 | 116.3 | 14.5 | 85 | 155.9 | 20.5 |
| 36 | 40.4 | 11.4 | 32 | 81 | 15.8 | 36 | 117 | 9.5 | 86 | 156.2 | 18 |
| 37 | 41 | 9.7 | 33 | 81.5 | 17.8 | 37 | 117.5 | 13.7 | 87 | 158.2 | 23.1 |
| 38 | 42 | 9.4 | 34 | 82.2 | 15 | 38 | 118 | 11.2 | 88 | 159 | 21.3 |
| 39 | 43 | 8.8 | 35 | 83.9 | 9.8 | 39 | 119 | 11.6 | 89 | 159.9 | 22.5 |
| 40 | 44 | 8.6 | 36 | 84.8 | 13.5 | 40 | 119.9 | 15.7 | 90 | 161 | 16 |
| 41 | 44.1 | 10.5 | 37 | 85.6 | 14 | 41 | 120.5 | 16.2 | 91 | 162 | 21.1 |
| 42 | 45.8 | 8.2 | 38 | 85.9 | 15.9 | 42 | 120.8 | 15.4 | 92 | 163 | 20.1 |
| 43 | 46 | 10 | 39 | 87 | 12.7 | 43 | 122 | 14.5 | 93 | 164 | 24.1 |
| 44 | 47 | 9 | 40 | 87.3 | 14 | 44 | 123 | 13.5 | 94 | 165 | 30.8 |
| 45 | 48 | 8.4 | 41 | 87.6 | 15 | 45 | 123.9 | 14.5 | 95 | 168 | 27.4 |
| 46 | 49.3 | 8.3 | 42 | 88 | 16.9 | 46 | 124 | 18 | 96 | 169 | 28.5 |
| 47 | 50 | 8.8 | 43 | 88.5 | 18.7 | 47 | 124.5 | 17.5 | 97 | 169.9 | 30.5 |
| 48 | 51 | 6.9 | 44 | 90 | 34 | 48 | 125 | 17 | 98 | 172.5 | 30.7 |
| 49 | 53 | 11.3 | | | | 49 | 125.5 | 18.7 | 99 | 173 | 28.4 |

| | IP | S. NO. | ANGLE ^o | DB | S. NO. | ANGLE ^o | DB | |
|-----|-------|--------|--------------------|-------|--------|--------------------|-----|------|
| 101 | 175 | 30.3 | 42 | 225.5 | 22.5 | 5 | 314 | 9 |
| 102 | 178 | 32.7 | 43 | 226 | 25 | 6 | 316 | 13.5 |
| 103 | 180 | 33 | 44 | 228 | 24.7 | 7 | 318 | 10.5 |
| 104 | 180.5 | 32.3 | 45 | 229 | 20.5 | 8 | 322 | 8 |
| 105 | 181 | 34.6 | 46 | 231 | 23 | 9 | 324 | 10.8 |
| 106 | 182 | 39.5 | 47 | 232 | 25 | 10 | 325 | 10.5 |
| 107 | 183 | 40 | 48 | 233 | 25.5 | 11 | 326 | 9.5 |
| 108 | 184 | 41 | 49 | 236 | 21.5 | 12 | 329 | 9.7 |
| 109 | 180 | 41 | 50 | 238 | 20.5 | 13 | 330 | 13.5 |
| 110 | 181 | 37.1 | 51 | 240 | 22 | 14 | 332 | 8.5 |
| 111 | 182 | 32.3 | 52 | 242 | 23.5 | 15 | 333 | 11.5 |
| 112 | 183.9 | 33.1 | 53 | 243 | 22.5 | 16 | 335 | 11 |
| 113 | 184.5 | 31.1 | 54 | 246 | 21.8 | 17 | 337 | 11.5 |
| 114 | 186 | 27.8 | 55 | 248 | 22.8 | 18 | 338 | 12.5 |
| 115 | 187 | 28.1 | 56 | 250 | 25.5 | 19 | 340 | 16.8 |
| 116 | 187 | 35.2 | 57 | 252 | 22.8 | 20 | 341 | 16.7 |
| 117 | 189 | 33.6 | 58 | 254 | 22 | 21 | 342 | 15.5 |
| 118 | 190 | 29.5 | 59 | 256 | 22.6 | 22 | 345 | 14.9 |
| 119 | 193 | 28.6 | 60 | 259 | 18.9 | 23 | 347 | 14 |
| 120 | 194 | 28 | 61 | 262 | 18.5 | 24 | 349 | 14.5 |
| 121 | 195 | 32.5 | 62 | 265 | 18 | 25 | 352 | 11.5 |
| 122 | 196 | 28.8 | 63 | 267 | 15.6 | 26 | 354 | 13 |
| 123 | 199 | 22.4 | 64 | 268 | 17.5 | 27 | 357 | 25 |
| 124 | 200 | 29.6 | 65 | 270 | 34.5 | 28 | 359 | 28.3 |
| 125 | 200.5 | 25 | | | | | | |
| 126 | 201.5 | 25.2 | 0 | 270 | 34.5 | | | |
| 127 | 202 | 21.5 | 1 | 272 | 20.5 | | | |
| 128 | 203 | 25.2 | 2 | 274 | 17.5 | | | |
| 129 | 204 | 25.2 | 3 | 276 | 14.5 | | | |
| 130 | 205 | 25.6 | 4 | 277 | 15.5 | | | |
| 131 | 206 | 22.3 | 5 | 279 | 16.5 | | | |
| 132 | 207 | 20.5 | 6 | 281 | 19.5 | | | |
| 133 | 208 | 21.2 | 7 | 284 | 15.5 | | | |
| 134 | 210 | 26.3 | 8 | 285 | 17.5 | | | |
| 135 | 212 | 24 | 9 | 290 | 15.0 | | | |
| 136 | 213.9 | 19.5 | 10 | 292 | 15.5 | | | |
| 137 | 214 | 17.1 | 11 | 293 | 16.5 | | | |
| 138 | 215 | 18 | 12 | 294 | 17 | | | |
| 139 | 216.1 | 20.7 | 13 | 295 | 13.5 | | | |
| 140 | 217 | 21 | 14 | 296 | 12.5 | | | |
| 141 | 219 | 20.1 | 15 | 297 | 13 | | | |
| 142 | 220 | 21.9 | 16 | 300 | 10.8 | | | |
| 143 | 220.2 | 21.7 | 17 | 302 | 16 | | | |
| 144 | 221 | 21.6 | 18 | 304 | 31 | | | |
| 145 | 221.9 | 22 | | | | | | |
| 146 | 222 | 22 | 0 | 304 | 31 | | | |
| 147 | 223 | 16.1 | 1 | 306 | 9.5 | | | |
| 148 | 223.5 | 17 | 2 | 307 | 5.5 | | | |
| 149 | 224 | 18.7 | 3 | 309 | 10.5 | | | |
| 150 | 225 | 20.6 | 4 | 312 | 13.5 | | | |