https://ntrs.nasa.gov/search.jsp?R=19690017341 2020-03-12T03:30:24+00:00Z



LIQUID-GAS INTERFACE RELATIONS IN INTERCONNECTED CONCENTRIC CYLINDER TANKAGE SYSTEMS

by Dr. F. W. Geiger Dr. J. C. May

R-244

IECHNICAL NOTE

CASE FILE COPY

September 1967

RESEARCH LABORATORIES BROWN ENGINEERING COMPANY, INC. HUNTSVILLE, ALABAMA

TECHNICAL NOTE R-244

LIQUID-GAS INTERFACE CONFIGURATIONS IN INTERCONNECTED CONCENTRIC CYLINDER TANKAGE SYSTEMS

September 1967

Prepared For

PROPULSION DIVISION PROPULSION AND VEHICLE ENGINEERING LABORATORY GEORGE C. MARSHALL SPACE FLIGHT CENTER

Prepared By

RESEARCH LABORATORIES ADVANCED SYSTEMS AND TECHNOLOGIES GROUP BROWN ENGINEERING, A TELEDYNE COMPANY HUNTSVILLE, ALABAMA

Contract No. NAS8-20073

By

Dr. F. W. Geiger Dr. J. C. May

ABSTRACT

A computer program has been developed to graphically and analytically define the static shape of the liquid-gas interface in the annular region between any two concentric cylinders in an axial force field for any Bond number and contact angle. The program also includes as a subroutine a previously reported method of calculating the static shape of the liquid-vapor interface within a single cylindrical tank. Static fluid surface coordinates for both the annular and central regions are given for Bond numbers (with inner cylinder radius as characteristic length) ranging from 10 to 500, for a contact angle of 5 degrees, and for a radius ratio of 1.5.

Approved:

E. J. Rodgers

Manager Mechanics and Thermodynamics Department

Approved:

R. C. Watson, Jr Vice President

TABLE OF CONTENTS

Page	
------	--

INTRODUCT	ION	1
STATEMENT	F OF THE PROBLEM	2
METHOD OF	ANALYSIS	4
Funda	amentals of Capillary Hydrostatics	4
Nondi Diffe Surfa	imensional Forms of the General rential Equation for Axisymmetric ces in Annular Regions	7
Nume Equat	tion	10
RESULTS OF	F CALCULATIONS	14
REFERENCE	ES	29
APPENDIX.	FORTRAN IV COMPUTER PROGRAM FOR DEFINING THE STATIC SHAPE OF SURFACES IN THE CENTRAL ANNULAR REGIONS OF CONCENTRIC CYLINDER TANKAGE SYSTEMS	30

LIST OF FIGURES

Figure	Title	Page
1	Fluid in Concentric Cylinder Tankage System with Axis Parallel to the Effective Acceleration of Gravity	. 3
2	Static Surface $\frac{R_2}{R_1} = 1.5$, $B_0 = 10$, $\theta = 5^\circ$. 15
3	Static Surface $\frac{R_2}{R_1} = 1.5$, $B_0 = 25$, $\theta = 5^{\circ}$. 16
4	Static Surface $\frac{R_2}{R_1} = 1.5$, $B_0 = 50$, $\theta = 5^\circ$. 17
5	Static Surface $\frac{R_2}{R_1} = 1.5$, $B_0 = 75$, $\theta = 5^{\circ}$. 18
6	Static Surface $\frac{R_2}{R_1} = 1.5$, $B_0 = 100$, $\theta = 5^{\circ} \cdots \cdots \cdots$	• 19
7	Static Surface $\frac{R_2}{R_1} = 1.5$, $B_0 = 250$, $\theta = 5^{\circ}$	20
8	Static Surface $\frac{R_2}{R_1} = 1.5$, $B_0 = 500$, $\theta = 5^{\circ}$	21

LIST OF TABLES

Table	Title	Page
1	Coordinates of Surfaces $\frac{R_2}{R_1} = 1.5$, $B_0 = 10$, $\theta = 5^{\circ} \dots$. 22
2	Coordinates of Surfaces $\frac{R_2}{R_1} = 1.5$, $B_0 = 25$, $\theta = 5^{\circ} \dots$. 23
3	Coordinates of Surfaces $\frac{R_2}{R_1} = 1.5$, $B_0 = 50$, $\theta = 5^{\circ}$. 24
4	Coordinates of Surfaces $\frac{R_2}{R_1} = 1.5$, $B_0 = 75$, $\theta = 5^{\circ}$. 25
5	Coordinates of Surfaces $\frac{R_2}{R_1} = 1.5$, $B_0 = 100$, $\theta = 5^\circ \dots$. 26
6	Coordinates of Surfaces $\frac{R_2}{R_1} = 1.5$, $B_0 = 250$, $\theta = 5^\circ$. 27
7	Coordinates of Surfaces $\frac{R_2}{R_1} = 1.5$, $B_0 = 500$, $\theta = 5^{\circ} \dots$	• 28

LIST OF SYMBOLS

Symbols	Definition
А	A constant, $lbm/ft-sec^2$
Bo	Bond number $\left(= \rho g R_1^2 / T \right)$
g	Effective acceleration of gravity, ft/sec^2
Н	Dimensionless modified y-coordinate of surface (\hbar/R_1)
h	y-coordinate of surface of fluid, ft
ħ	Modified y-coordinate of surface, ft
Р	Pressure of vapor and gas above liquid, lbm/ft-sec ²
р	Static pressure in liquid, lbm/ft-sec ²
R ₁	Radius of inside cylinder, ft
R ₂	Radius of outside cylinder, ft
R_{c1}, R_{c2}	Principal radii of curvature of surface at a point, ft
r	Radial distance (from axis of symmetry), ft
S	Nondimensional arc length in H, t-plane
Т	Surface tension of liquid, lbm/sec ²
t	Nondimensional radial distance (r/R_1)
У	Distance perpendicular to r-direction, ft
α	Inclination of surface in H, t-plane, radians
ρ	Density of fluid (liquid), lbm/ft ³
θ	Contact angle

LIST OF SYMBOLS (Continued)

Subscripts	Definition
а	Of central region
Ъ	Of annular region
i	Value at end of ith step of integration process
m	Mean value
r	Derivative with respect to r
S	Value at surface
t	Derivative with respect to t
w	Value at inner wall of annulus

INTRODUCTION

For the smooth, reliable, and consistent operation of a vehicle propulsion system, only liquid must be delivered to the engine. Therefore, the liquid phase must always be located at the outlet of the propellant tank. Clodfelter¹, recognizing that the pressure drop across a liquidgas interface could be utilized for mass transfer, suggested that tanks consisting of several properly sized, interconnected, concentric cylinders be used to position liquid propellants at low gravitational accelerations. In an extension of the concepts advanced by Clodfelter, the present report utilizes the mathematical techniques developed by Bashforth and Adams² and Geiger^{3, 4} to predict the static shapes of axially symmetric liquid surfaces within concentric cylinder tankage systems at desired Bond numbers and contact angles. The method is used to predict the surface shapes for a radius ratio (outer to inner cylinder) of 1.5, for a contact angle of 5 degrees, and for Bond numbers from 10 to 500 (using the inner radius as the characteristic length).

An alternate analysis of the annular region has been given by Seebold et al⁵, who numerically integrated one of the differential equations describing the surface using the Adams predictor-corrector method. Their results are plotted, but the accuracy with which the graphs can be read is not great. The present method can be expected to yield considerably more accurate results.

1

STATEMENT OF THE PROBLEM

The problem can be defined in consecutive steps as follows: Consider a section of an interconnected concentric cylinder tankage system filled with fluid (Figure 1). Select a y-axis along the axis of symmetry of the system and an r-axis perpendicular to the y-axis in the radial direction. Let the effective acceleration of gravity, g, act in the minus y-direction. Let the mean height of the fluid in the annular region be h_{mb} and the mean height of the fluid in the annular region be h_{mb} and the mean height of the fluid in the central region be h_{ma} . Let the height of the liquid at an arbitrary r be h(r). Assume that liquid density, surface tension, contact angle, and pressure above the liquid are constant throughout the system. Find the h- and rcoordinates of the liquid-vapor interface in each region of the system.



Figure 1. Fluid in Concentric Cylinder Tankage System with Axis Parallel to the Effective Acceleration of Gravity

METHOD OF ANALYSIS

FUNDAMENTALS OF CAPILLARY HYDROSTATICS

The basic equation of capillarity, sometimes referred to as the Young and Laplace equation, is

$$P_0 - P_s = T\left(\frac{1}{R_{c1}} + \frac{1}{R_{c2}}\right)$$
 (1)

where

$P_0 - P_s$	-	the difference in pressure across a liquid-gas interface at any point of the interface
R _{c1} , R _{c2}	-	the principal radii of curvature of the interface at that point
\mathbf{P}_{0}	-	the pressure in the gas at the interface
Ps	-	the pressure in the liquid at the interface
т	-	the surface tension of the liquid

Whenever a liquid-gas interface is axisymmetric, R_{c1} and R_{c2} can be expressed explicitly. Thus,

$$P_{0} - P_{s} = T \left[\frac{h_{rr}}{\left(1 + h_{r}^{2}\right)^{3/2}} + \frac{h_{r}}{r \left(1 + h_{r}^{2}\right)^{1/2}} \right]$$

$$= \frac{T}{r} \frac{d}{dr} \left[\frac{rh_{r}}{\left(1 + h_{r}^{2}\right)^{1/2}} \right]$$
(2)

where h(r) is the ordinate of the liquid-gas interface and the subscript r indicates the derivative with respect to r.

In a previous report, Geiger⁴ used the procedure which follows to relate the hydrostatic pressure, p = p(r, y), at any point in a fluid to the pressures, P_0 and P_s , on either side of the liquid-gas interface for a single cylinder of radius R_1 . Here the procedure is used for the case of two concentric cylinders of radius R_1 and R_2 .

First the pressure, p, at any point, (r, y), is expressed as follows (see Figure 1):

$$P + \rho g y = P_0 + A \tag{3}$$

where A is a constant to be determined. Thus,

$$P_{s} + \rho g h = P_{0} + A \tag{4}$$

 P_s is now eliminated from Equations 2 and 4, yielding

$$\frac{T}{r} \frac{d}{dr} \left[\frac{r h_r}{\left(1 + h_r^2 \right)^{1/2}} \right] = \rho g h - A$$
(5)

Equation 5 is integrated after multiplying it by r to give

$$T \left[\frac{r h_{r}}{\left(1 + h_{r}^{2} \right)^{\frac{1}{2}}} \right] \stackrel{R_{2}}{=} \rho g \int_{R_{1}}^{R_{2}} dr - A \int_{R_{1}}^{R_{2}} r dr \qquad (6)$$

where R_1 and R_2 are any two radii.

For the annular region between concentric cylinders of radii R_1 and R_2 (see Figure 1),

$$h_r = -\cot \theta \ at R_1$$

 $h_r = +\cot \theta \ at R_2$

where θ is the contact angle between the liquid-gas interface and the tank wall (measured in the fluid). Therefore,

$$T \left[\frac{r h_r}{\left(l + h_r^2 \right)^{1/2}} \right] \stackrel{R_2}{=} T \left(R_2 + R_1 \right) \cos \theta \qquad (7)$$

Now

$$\int_{R_{1}}^{R_{2}} r \, dr = \frac{R_{2}^{2} - R_{1}^{2}}{2}$$
(8)

Let V be the volume of fluid above the r-axis. Then

$$V = 2\pi \int_{R_{1}}^{R_{2}} h r dr = \pi (R_{2}^{2} - R_{1}^{2}) h_{mb}$$
(9)

where h_{mb} is the average height of the interface above the r-axis. The above results are then substituted into Equation 6, giving

$$T \left(R_{2} + R_{1}\right) \cos \theta = \rho g \left(\frac{R_{2}^{2} - R_{1}^{2}}{2}\right) h_{mb} - A \left(\frac{R_{2}^{2} - R_{1}^{2}}{2}\right)$$
(10)

from which it follows that

$$A = \rho g h_{mb} - \frac{2T \cos \theta}{(R_2 - R_1)}$$
(11)

The pressure at any point in the fluid is now written

$$P + \rho g y = P_0 + \rho g h_{mb} - \frac{2T \cos \theta}{R_2 - R_1}$$
 (12)

which defines the pressure at all points in the annular region. For a single cylindrical tank, the expression for the pressure at any point can be written 4

$$P + \rho g y = P_0 + \rho g h_{ma} - \frac{2T \cos \theta}{R_1}$$
(13)

where R_1 is the radius of the cylinder. This completes the procedure.

Equations 12 and 13 therefore define respectively the pressure in the annular and central regions of any concentric cylinder system. At the same y, p is the same and

$$h_{mb} - h_{ma} = \frac{2T \cos \theta}{\rho g} \left(\frac{1}{R_2 - R_1} - \frac{1}{R_1} \right)$$
(14)

when

$$R_2 = 2R_1$$
, $h_{mb} - h_{ma} = 0$
 $R_2 > 2R_1$, $h_{mb} - h_{ma} < 0$
 $R_2 < 2R_1$, $h_{mb} - h_{ma} > 0$

Thus the relative importance of the radii in a concentric cylinder tankage system in positioning the fluid in one region with respect to its position in the other becomes apparent.

NONDIMENSIONAL FORMS OF THE GENERAL DIFFERENTIAL EQUATION FOR AXISYMMETRIC SURFACES IN ANNULAR REGIONS

The general differential equation of capillary hydrostatics for axisymmetric surfaces can be obtained from Equations 5 and 11. Thus

$$\frac{\mathrm{T}}{\mathrm{r}} \frac{\mathrm{d}}{\mathrm{d}\mathrm{r}} \left[\frac{\mathrm{r} \mathrm{h}_{\mathrm{r}}}{\left(1 + \mathrm{h}_{\mathrm{r}}^{2} \right)^{1/2}} \right] = \rho \mathrm{g} \mathrm{h} - \rho \mathrm{g} \mathrm{h}_{\mathrm{mb}} + \frac{2\mathrm{T} \cos \theta}{\mathrm{R}_{2} - \mathrm{R}_{1}} \qquad (15)$$

Letting

$$\hbar = h - h_{mb} + \frac{2T \cos \theta}{\rho g (R_2 - R_1)} ,$$

one obtains

$$\frac{\mathrm{T}}{\mathrm{r}} \frac{\mathrm{d}}{\mathrm{dr}} \left[\frac{\mathrm{r} \, \tilde{\mathrm{h}}_{\mathrm{r}}}{\left(1 + \mathrm{h}_{\mathrm{r}}^{2} \right)^{1/2}} \right] = \rho \, \mathrm{g} \, \tilde{\mathrm{h}} \quad . \tag{16}$$

The \hbar and r coordinates are made nondimensional here by dividing by the radius of the inside cylinder, R_1 . Actually, this amounts to taking the quantity R_1 as the unit of length. For the sake of simplicity, the following transformations have been made:

$$\frac{r}{R_1} = t$$
, $\frac{\hbar}{R_1} = H$, $B_0 = \frac{\rho g R_1^2}{T}$

where the Bond number, B_0 , is the ratio of the body forces to capillary forces in the prevailing force field. The dimensionless differential equation of the equilibrium surface profile can now be written

$$\frac{1}{t} \frac{t}{dt} \left[\frac{t H_t}{\left(1 + H_t^2\right)^{1/2}} \right] = B_0 H$$
(17)

For the special case of zero Bond number, the reader is referred to a report by Clodfelter¹ in which numerical solutions for the shape of the liquid-vapor interface in the annular region between concentric cylinders are given.

The nonlinearity of Equation 17 renders general closed-form solution impossible. However, various schemes have been devised by Geiger³, Seebold et al⁵, and Bashforth and Adams² to integrate this equation or an equivalent equation numerically. The technique which Geiger proposed will be used in this report. This involves further transformations.

Let

$$H_t = tan \alpha$$

where α is the angle between the H-to-t (or the h-to-r) curve and the t (or r) axis. Equation 17 then can be written

$$\frac{1}{t}\frac{d}{dt}\left(t\sin\alpha\right) = B_{0}\left(H_{w} + \int_{1}^{t}\tan\alpha\,dt\right)$$

or

$$\frac{\sin \alpha}{t} + \cos \alpha \frac{d\alpha}{dt} = B_0 \left(H_w + \int_1^t \tan \alpha \, dt \right)$$
(18)

where H_w is the undetermined value of H at the wall at which t = 1 or $r = R_1$. The boundary conditions are

at
$$t = 1$$
: $\alpha = -\left(\frac{\pi}{2} - \theta\right)$
at $t = \frac{R_2}{R_1}$: $\alpha = \frac{\pi}{2} - \theta$

for any contact angle. Although Equation 18 can be integrated numerically, it is difficult to use for low contact angles (where α is large). This difficulty can be avoided by changing the independent variable from t to s, the arc length in the H-to-t plane. This technique was first used by Bashforth and Adams² although they used a surface curvature as characteristic length (i.e., in place of R_1). Thus,

$$\frac{dt}{ds} = \cos \alpha ,$$

$$\int_{1}^{t} dt = \int_{0}^{s} \cos \alpha \, ds$$

where s = 0 at t = 1, and

$$t = 1 + \int_{0}^{s} \cos \alpha \, ds$$

Also,

$$\cos \alpha \frac{\mathrm{d}\alpha}{\mathrm{d}t} = \cos \alpha \frac{\mathrm{d}\alpha}{\mathrm{d}s} \frac{\mathrm{d}s}{\mathrm{d}t} = \frac{\mathrm{d}\alpha}{\mathrm{d}s}$$

Now, Equation 18 becomes

$$\frac{\sin \alpha}{1 + \int_{0}^{s} \cos \alpha \, ds} + \frac{d\alpha}{ds} = B_{0} \left(H_{w} + \int_{0}^{s} \sin \alpha \, ds \right)$$
(19)

s

A

where $\frac{d\alpha}{ds}$ is the curvature of the surface in the H-to-t plane.

NUMERICAL INTEGRATION OF THE DIFFERENTIAL EQUATION

Equation 19 is to be integrated numerically to obtain the shape of the interface in the annular region between concentric cylinders. A subinterval of s is chosen of length $\Delta s = \frac{1}{n}$, where n is a large number; e.g., 10,000. Starting at s = 0, a forward difference scheme is used to calculate the values of α , t, and (h - h_{ma})/R₁ at the end of each successive subinterval.

Since $\frac{d\alpha}{ds}\Big|_{s=0}$ and H_w are unknown (they are interrelated

through Equation 20), an iterative procedure must be employed to determine that value of $\frac{d\alpha}{ds} \Big|_{s=0}$ which satisfies the boundary conditions. A minimum possible value of $\frac{d\alpha}{ds} \Big|_{s=0}$ is easily determined. From

Equation 19 it follows that

$$\frac{d\alpha}{ds}\Big|_{s=0} = B_0 H_w - \sin\alpha \Big|_{s=0}$$

$$= B_0 H_w - \sin\left[\theta - \frac{\pi}{2}\right]$$

$$= B_0 H_w + \cos\theta$$

$$= B_0 \frac{h_w - h_{mb}}{R_1} + \frac{2R_1 \cos\theta}{R_2 - R_1} + \cos\theta$$

$$= B_0 \frac{h_w - h_{mb}}{R_1} + \frac{R_1 + R_2}{R_2 - R_1} \cos\theta \qquad (20)$$

Since $h_w - h_{mb}$ is positive for acute values of θ and is negative for obtuse values of θ , it is clear that

$$\left|\frac{\mathrm{d}\alpha}{\mathrm{d}s}\right|_{s=0} > \left|\frac{\mathrm{R}_{1} + \mathrm{R}_{2}}{\mathrm{R}_{2} - \mathrm{R}_{1}}\cos\theta\right|$$

or that the latter is the minimum sought. Unfortunately, no convenient maximum value for $\left| \frac{d\alpha}{ds} \right|_{s = 0}$ has been found. As an initial or trial value of $\left| \frac{d\alpha}{ds} \right|_{s = 0}$, an arbitrary value is

selected somewhat larger in magnitude than the minimum possible value. The corresponding value of H_w is found from Equation 20. Then values of α , H, t, and $\frac{d\alpha}{ds}$ at the end of each subinterval, Δs , are calculated. In making these calculations, the following approximations are made:

$$\alpha_{i+1} = \alpha_i + \frac{d\alpha}{ds} \Big|_i \Delta s$$

$$H_{i+1} = H_i + \left(\frac{\sin \alpha_i + \sin \alpha_i + 1}{2}\right) \Delta s$$

$$t_{i+1} = t_i + \left(\frac{\cos \alpha_i + \cos \alpha_i + 1}{2}\right) \Delta s \quad ; \quad (21)$$

and $\frac{d\alpha}{ds}\Big|_{i+1}$ is calculated from

$$\frac{\mathrm{d}\alpha}{\mathrm{d}s}\Big|_{i+1} = B_0 H_{i+1} - \frac{\sin\alpha_{i+1}}{t_{i+1}} . \qquad (22)$$

When
$$t = 1 + \int_{0}^{\infty} \cos \alpha \, ds$$
 becomes equal to $\frac{\pi}{R_1}$, $\alpha(s)$ must equal $\frac{\pi}{2} - \theta$ or be very close to it. If it is not, a new value of $\frac{d\alpha}{ds} \bigg|_{s = 0}$

must be selected and the procedure repeated. In the iterative technique used to obtain $\frac{d\alpha}{ds}\Big|_{s = 0}$ for acute contact angles, it should be noted that:

1. If
$$1 + \int_{0}^{s} \cos \alpha \, ds < \frac{R_2}{R_1}$$
 for all s, $\frac{d\alpha}{ds}\Big|_{s=0}$ is too large.

2. If at
$$1 + \int_{0}^{s} \cos \alpha \, ds = \frac{R_2}{R_1}, \frac{\pi}{2} - \alpha(s) < \theta, \frac{d\alpha}{ds} \Big|_{s = 0}$$

is too large.
3. If at
$$1 + \int_{0}^{s} \cos \alpha \, ds = \frac{R_2}{R_1}, \frac{\pi}{2} - \alpha(s) > \theta, \frac{d\alpha}{ds} \bigg|_{s = 0}$$

is too small. Similar conditions can be written for obtuse contact angles.

Once $\frac{d\alpha}{ds}\Big|_{s=0}$ is known, the surface coordinates $\frac{h-h_{ma}}{R_1}$ and $t\left(\frac{r}{R_1}\right)$ are also known. In fact, $t = 1 + \int_0^s \cos \alpha \, ds$ $\frac{h-h_{ma}}{R_1} = \frac{h-h_{mb}}{R_1} + \frac{h_{mb}-h_{ma}}{R_1}$ $= \frac{1}{B_0}\left(\frac{d\alpha}{ds}\Big|_{s=0} - \cos \theta\right) - \frac{2}{B_0} - \frac{R_1}{R_2 - R_1} \cos \theta$ $+ \int_0^s \sin \alpha \, ds + \frac{2}{B_0} \cos \theta \left(\frac{R_1}{R_2 - R_1} - 1\right)$ $= \frac{1}{B_0}\left(\frac{d\alpha}{ds}\Big|_{s=0} - 3\cos \theta\right) + \int_0^s \sin \alpha \, ds$. (23) A listing of the FORTRAN IV computer program which was used to obtain surface profiles (values of $\frac{h - h_{ma}}{R_1}$ and the corresponding values of $t = \frac{r}{R_1}$) for both the central and annular regions of concentric cylinder tankage systems is given in the Appendix.

RESULTS OF CALCULATIONS

Computer calculations were made for $\frac{R_2}{R_1} = 1.5$, for seven Bond numbers between 10 and 500, and for a contact angle of 5 degrees. For these calculations, an arbitrary value of Δs , the increment of arc, was selected for the central region and the value of Δs for the annular region was chosen such that the error in the contact angle was less than 0.0005 radians or 0.0286 degrees.

The results of the calculations are plotted in Figures 2 through 8. Selected results are given in Tables 1 through 7. In the tables the last point is that point for the minimum ordinate in the annular region for which the results were printed by the computer. Results were printed for every 50th calculated point in this region. The results show the expected trends.

Seebold et al⁵ numerically integrated one of the differential equations describing the surface in the annular region using the Adams predictor-corrector method. They used the larger radius as characteristic length (inner radius is used in this report), and their maximum Bond number was 30 (the corresponding Bond number of this report would be 13.3). They plotted height at the outer wall, maximum depression, and height at the inner wall against radius ratio for various Bond numbers for contact angles of 0, 5, and 15 degrees.

When the results of the present study are compared with those just discussed (and this can only be done for the lowest Bond number of this report, 10, and for the one radius ratio), a difference in $\frac{h - h_{mb}}{R_2}$ appears which is of the order of magnitude of ±0.01; and this number is too large to be explained on the basis of errors in interpolation and in curve reading, estimated as ±0.004. Thus the agreement between the two sets of results cannot be said to be good.

To what this is attributable is not known. It is believed, however, that the present results should be correct to within about 0.0001 in $(h - h_{ma})/R_1$ or to within 0.00007 in $(h - h_{mb})/R_2$.















COORDINATES OF SURFACES

$\frac{R_2}{R_1} =$	1.5	$B_0 = 10$	$\theta = 5^{\circ}$

 $\Delta s = 2.5 \times 10^{-4}$ (annular region)

Central Region		Annular Region	
	h - h _{ma}		h – h _{ma}
t		t	R ₁
0.00000	-0.13601	1.00000	0.32979
0.12497	-0.13352	1.01174	0.28164
0.24971	-0.12575	1.03670	0.23849
0.37392	-0.11179	1.07185	0.20310
0.49698	-0.09004	1.11417	0.17667
0.61775	-0.05801	1.16105	0.15953
0.73398	-0.01227	1.21034	0.15159
0.84124	0.05154	1.26027	0.15254
0.93125	0.13774	1.30930	0.16203
0.98961	0.24746	1.35600	0.17970
1.00002	0.30111	1.39894	0.20519
		1.43657	0.23801
h h		1.46717	0.27744
$\frac{mb}{R_1}$	= 0.19924	1.48888	0.32237
		1.49980	0.37104
		1.50001	0.37328
		1.23533	0.15098

COORDINATES OF SURFACES

$\Delta s = 2.5 \times 10^{-4}$ (annular region)

Central Region

Annular Region

	h - h _{ma}		h - h _{ma}
t	R1	t	R ₁
0.00000	-0.07220	1.00000	0.19996
0.12500	-0.07145	1.01345	0.15235
0.24997	-0.06897	1.04195	0.11150
0.37487	-0.06401	1.08053	0.07990*
0.49954	-0.05505	1.12523	0.05769
0.62351	-0.03925	1.17332	0.04426
0.74534	-0.01168	1.22298	0.03885
0.86066	0.03590	1.27289	0.04094
0.95657	0.11483	1.32195	0.05032
1.00000	0.21716	1.36900	0.06708
		1.41254	0.09152
		1.45051	0.12391
h _{mb} - h _{ma}	= 0.07970	1.48007	0.16407
κı		1.49755	0.21072
		1.50001	0.22753
		1.23547	0.03868

COORDINATES OF SURFACES

$$\frac{R_2}{R_1} = 1.5$$
 $B_0 = 50$ $\theta = 5^{\circ}$

$\Delta s = 1.25 \times 10^{-4}$ (annular region)

Centra	l Region	Annular Region	
t	$\frac{11 - 11_{ma}}{R_1}$	t	$\frac{11 - 11}{R_1}$
0.00000	-0.03907	1.00000	0.14712
0.12500	-0.03891	1.01575	0.10025
0.25000	-0.03834	1.04849	0.06278
0.37499	-0.03698	1.09065	0.03614
0.49995	-0.03395	1.13753	0.01897
0.62476	-0.02723	1.18658	0.00949
0.74880	-0.01220	1.23645	0.00644
0.86894	0.02136	1.28635	0.00924
0.96986	0.09290	1.33554	0.01797
1.00000	0.16392	1.38307	0.03333
		1.42727	0.05653
h h		1.46519	0.08890
$\frac{\text{mb}}{\text{R}_1}$	= 0.03985	1.49186	0.13091
		1.50000	0.16404
		1.23645	0.00644

COORDINATES OF SURFACES

 $\frac{R_2}{R_1} = 1.5$ $B_0 = 75$ $\theta = 5^{\circ}$

$\Delta s = 2.5 \times 10^{-4}$ (annular region)

Centra	1 Region	Annular Region	
+	h - h _{ma}	+	$\frac{h - h_{ma}}{R}$
0.00000	-0.02642	<u> </u>	0.12440
0.12500	-0.02638	1.01746	0.07841
0 .250 00	-0.02620	1.05312	0.04374
0.37500	-0.02572	1.09744	0.02087
0.49999	-0.02443	1.14545	0.00711
0.62494	-0.02098	1.19492	0.00005
0.74955	-0.01160	1.24487	-0.00177
0.87168	0.01395	1.29476	0.00111
0.97564	0.08028	1.34411	0.00901
1.00002	0.13795	1.39203	0.02310
		1.43671	0.04534
		1.47431	0.07801
h _{mb} - h _m	$\frac{a}{2} = 0.02657$	1.49747	0.12186
к1		1.50002	0.13663
		1.24487	-0.00177

COORDINATES OF SURFACES

 $\frac{R_2}{R_1} = 1.5$ $B_0 = 100$ $\theta = 5^\circ$

$\Delta s = 2.50 \times 10^{-4}$ (annular region)

Central Region		Annular Region		
	h - h _{ma}		h – h _{ma}	
t	R ₁	<u>t</u>	R1	
0.00000	-0.01989	1.00000	0.11056	
0.12500	-0.01987	1.01892	0.06530	
0.25000	-0.01981	1.05677	0.03306	
0.37500	-0.01961	1.10251	0.01315	
0.49999	-0.01899	1.15118	0.00188	
0.62498	-0.01702	1.20086	-0.00355	
0.74980	-0.01073	1.25083	-0.00467	
0.87295	0.00956	1.30075	-0.00193	
0.97903	0.07186	1.35021	0.00518	
1.00002	0.12159	1.39845	0.01818	
		1.44348	0.03966	
h _{mb} - h _{ma} R ₁		1.48059	0.07279	
	= 0.01992	1.49983	0.11828	
		1.50000	0.12002	
		1.23834	-0.00475	

COORDINATES OF SURFACES

 $\frac{R_2}{R_1} = 1.5$ $B_0 = 250$ $\theta = 5^{\circ}$

 $\Delta s = 1.0 \times 10^{-4}$ (annular region)

Central Region h - h		Annular Region h - h		
t	$\frac{11}{R_1}$	t	n nma R1	
0.00000	-0.00797	1.00000	0.07520	
0.10000	-0.00797	1.01730	0.04010	
0.20000	-0.00797	1.05027	0.01788	
0.30000	-0.00797	1,08829	0.00569	
0.40000	-0.00796	1.12776	-0.00069	
0.49999	-0.00794	1.16762	-0.00391	
0.59999	-0.00783	1.20759	-0.00538	
0.69998	-0.00734	1.24758	-0.00575	
0.79995	-0.00513	1.28757	-0.00521	
0.89933	0.00501	1.32753	-0.00356	
0.98693	0.04898	1.36738	-0.00016	
1.00000	0.07995	1.40684	0.00629	
		1.44496	0.01820	
h _{mb} – h _m	la _ 0 00707	1.47867	0.03937	
R ₁	- = 0.00797	1.49908	0.07305	
		1,50001	0.07918	
		1.24258	-0.00576	

COORDINATES OF SURFACES

$\frac{R_2}{R_1} = 1.5$	$B_0 = 500$	$\theta = 5^{\circ}$
-------------------------	-------------	----------------------

$\Delta s = 1.0 \times 10^{-4}$ (annular region)

Central Region		Annular	Annular Region		
t	$\frac{h - h_{ma}}{R_1}$	t	h - h _{ma} R1		
0.0000	-0.00398	1.00000	0.05526		
0.10000	-0.00398	1.02102	0.02279		
0.20000	-0.00398	1.05752	0.00697		
0.30000	-0.00398	1.09694	0.00043		
0.40000	-0.00398	1.13684	-0.00220		
0.49999	-0.00398	1.17682	-0.00324		
0.59999	-0.00398	1.21682	-0.00362		
0.69998	-0.00392	1.25681	-0.00367		
0.79998	-0.00345	1.29681	-0.00346		
0.89985	0.00075	1.33680	-0.00281		
0.99082	0.03591	1.37676	-0.00120		
1.00001	0.05772	1.41656	0.00269		
		1.45538	0.01201		
h _{mb} - h _{ma}	- 0 00309	1.49870	0.05040		
R ₁	= 0.00398	1.50001	0.05727		
		1.24181	-0.00368		

REFERENCES

- Clodfelter, R. G., "Fluid Mechanics and Tankage Design for Low Gravity Environments", Air Force Aero-Propulsion Laboratory, Wright-Patterson Air Force Base, ASD-TDR-63-506, 1963
- Bashforth, F. and Adams, J. C., <u>An Attempt to Test the Theories</u> of Capillary Action by Comparing the Theoretical and Measured Forms of Drops of Fluid, Cambridge, England (at the University Press), 1883, Authorized reprint by University Microfilms, Inc., Ann Arbor, Michigan
- Geiger, F. W., "Hydrostatics of a Fluid in a Cylindrical Tank at Low Bond Numbers", Brown Engineering Company, Inc., Technical Note R-207, July 1966
- Geiger, F. W., "Axially Symmetric Static Surfaces for Bond Numbers 10-1000 and for Contact Angles of Zero and Five Degrees", Brown Engineering Company, Inc., Technical Note R-207A, October 1966
- 5. Seebold, J. G. et al, "Capillary Hydrostatics in Annular Tanks", Journal of Spacecraft and Rockets, Vol 4, No. 1, January 1967, pp. 101-105.

APPENDIX

FORTRAN IV COMPUTER PROGRAM FOR INTERFACE CONFIGURATIONS IN CONCENTRIC CYLINDER TANKAGE SYSTEMS

Cross-Reference Between Symbols (Input and Output) Main Program

Algebraic Symbol	FORTRAN Symbol	Description
α	ALPHA	Local angle of inclination of surface
$\alpha_{i-1} + \frac{d\alpha}{ds} + \frac{\Delta s}{2}$	ALPME	
$1 + \int_{0}^{s} \cos \alpha ds$	ANTCOS	$t = \frac{r}{R_1}$
$\int_{0}^{\cdot} \sin \alpha \mathrm{ds}$	ANTSIN	H - H _w
B ₀	BO	Bond number
$\frac{\mathrm{d}\alpha}{\mathrm{d}s}$	DADS	Local curvature of surface
$\frac{d\alpha}{ds}s = 0$	DADSO	$\frac{d\alpha}{ds}$ at inner wall
Δs	DS	Increment of arc
$\frac{h_{mb} - h_{ma}}{R_1}$	DIFF	Nondimensional difference between mean height s
h ma	HM	Mean height in annular region
H _w	HR 1	Value of H at inner wall
R ₁	R l	Inner radius

	Algebraic Symbol	FORTRAN Symbol	Description
R ₂		R 2	Outer radius
$\frac{R_2}{R_1}$		OLIM	Ratio of outer to inner radius
π		PI	3.14159
θ		THET	Contact angle in degrees
θ		THETA	Contact angle in radians
θ		XXX	Contact angle in radians (for subroutine)

```
DIMENSION ALPHA(2),X(2000),Y(2000),K(2000),BCDX(12),BCDY(12)
    DIMENSION ARRAY(8)
    DIMENSION XX(2000), YY(2000)
     EQUIVALENCE (P1, PI)
    DATA BCD/6H /
    CALL CAMRAV(935)
     DO 1000 JJ = 1.12
     BCDX(1) = BCD
1000 \text{ BCUY(JJ)} = \text{BCD}
     CALL BC (BCDX(6), 6H R/R1)
     CALL BC (BCDY(6), 6H (H-HM)
     CALL BC (BCDY(7), 6H)/R1 )
     PI=3.1415927
     READ(5,5) THET , R1, R2
     OLIME R2/RL
     XXX = THET
     XXX = XXX + PI/180.
    THETA=THET /57.29578
     READ (5,96) KK
  96 FORMAT(14)
     D_0 488 N = 1.KK
     READ(5,5) 60
     IF(B0 = 100.) 401,401,402
 401 DS = .00025
     GO TO 403
```

```
402 DS = .00010
 403 CONTINUE
     Dx = 100.
     DY = (R2+R1)/(R2-R1) * COS(THETA)
     CALL FRED (BO, XXX, DS, XX, YY, JJ)
  105 HM = 0.0
      KOUNT=U
   5 FORMAT(3F10.0)
C** BO=RHO*G*R1**2/T
      WRITE(6,1011) BO, DS
 1011 FORMAT(1H ,10X, 3HB0=,1X, E15.8, E15.8)
      Dx = DX + 1.
      DY = DY - 1.
      DAUSO = DX
      DALPDS = DADSO
      IF(DS.LT. .001)LL=50
      IF (DS.LT..0001)LL=100
      IF(DS.LT..00001)LL=1000
   10 HRI = (DALPDS - SIN(P1/2 - THETA))/BO
      KOUNT=KOUNT + 1
      IF(KOUNT - 100) 50,30,30
   50 CONTINUE
      ANTCOS=1.0
      ANTSIN=0.0
      X(1)=1.0
```

```
Y(1) = 0.0
   K(1) = 1
   IK = 2
   II = 1
   I = 1
   ALPHA(1) = -(P1/2.-THETA)
   WEITE(6.6) DALPUS, HR1
5 FORMAT(1HD, SHDALPDS, 2X, F10.5, 3HHR1, 2X, F10.5/)
20 CONTINUE
   1=2
   ALPHA(I) = ALPHA(I-1) + DALPDS*DS
   ALPME = ALPHA(I-1) + DALPDS*US/2.
  ANTSIN = ANTSIN + DS/6.*(SIN(ALPHA(I))+4.*SIN(ALPME)+SIN(ALPHA(I-1))
  1)))
   SAVE= ANTCOS
  ANTCUS = ANTCOS + US/6.*(COS(ALPHA(I))+4.*COS(ALPME)+COS(ALPHA(I-1
  1)))
   IF(I1 - 11/LL*LL) 90,90,91
90 IK = IK + 1
91 CONTINUE
   X(IK) = ANTCUS
   Y(IK) = ANTSIN
   K(IK) = II
   II = II + 1
   IF (SAVE-ANTCOS) 51,51,22
```

```
51 CONTINUE
   DALPUS = BU*HR1 + BO*ANTSIN - SIN(ALPHA(1))/ANTCÓS
   IF (DALPDS) 23,23,29
29 CONTINUE
   ALPHA(1) = ALPHA(2)
   IF (ANTCOS - 0LIM) 20,52,52
52 CONTINUE
    WRITE(6,8) ANTCOS
 8 FORMAT(1HU,6HANTCOS, F10.5)
    WRITE(6,7) ALPHA(1)
 7 FORMAT(1H0,6HALPHA , F10.5)
    1F(ABS(ALPHA(1)+THETA-P1/2.) -.0005) 30,53,53
53 IF(ALPHA(I) - (P1/2.-THETA)) 23,30,22
22 DX = DX - (DX - DY)/2.
    IF (ABS(DADSO-DX)-.00000001) 101,101,103
101 \text{ US} = 05/2.
   GO TO 105
103 DAUSO = DX
   DALPDS = DX
    WRITE(6,75) ALPHA(2), ALPHA(1), X(IK), SAVE, Y(IK)
75 FORMAT(2X,10HALPHA(2) =, F10.5,2X,10HALPHA(1) =, F10.5,2X,7HT(II) =,
  1F10.5/2X/9HT(II-1) =/F10.5/2X/7HH(II) =/F10.5/)
   GO TU 10
23 DX = DX + (DX - DY)/2.
```

```
DY = DY + 2./3.*(DX-DY)
```

```
IF (ABS(DADSO-DX)-.00000001) 102,102,104
102 \text{ OS} = \text{DS/2.}
    GO TO 105
104 \text{ DADSO} = \text{DX}
    DALPDS = DX
    WRITE(6,75) ALPHA(2), ALPHA(1), X(IK), SAVE, Y(IK)
    GO TO 10
 30 CONTINUE
    DIFF = 2./BO * (R1/(R2-R1) - 1.) * COS(THETA)
    DO 666 I=1, IK
   Y(1) =
                         HM/R1 + 1./BO*(DADSO-COS(THETA)-2.*R1*COS(THE
   1TA)/(R2-R1)) + Y(I) + DIFF
666 CONTINUE
    ALPHA(2) = ALPHA(2) + 57.29578
    WRITE(6,66) DO, ALPHA(2), THET , DS
 66 FORMAT(11,10HBOND NO. =>F10.0/3X,7HALPHA =>F10.5/3X,7HTHETA =>F10.
   25,3X,9HARC LG. =, F10.6/)
   WRITE(0167)
 67 FORMAT( 5X,7HPT, NO.,7X,4HR/R1,7X,9H(H-HM)/R1/)
    WRITE(5,68) (K(J), X(J), Y(J), J = 1,IK)
 66 FORMAT(I10, 4X, F10.5, 4X, F10.5)
    YMIN = 1000.
    Y_{MAX} = -1000.
    Do 4000 II = 1.1K
    IF(Y(II).LT.YMIN) YMIN=Y(II)
```

```
IF(Y(II) G(YMAX) YMAX=Y(II)
4000 CONTINUE
    YB = -.5
    YT = 1.0
    XL = XX(1)
    XR = OLIM
    CALL SCUUTV
    WRITE(16,1600)
    180.
    1THET .
    IOLIM
              9
    1Y(1)
             1Y(IK)
              .
    1YMIN
               .
    105,
    1DIFF
1600 FORMAT(1H1,5X,12HBOND NO. = ,F10.5,//,5X,31HACTUAL CONTACT ANGLE
    1, THE TA = , F10.5, //, 5X, 9HR2/R1 = , F10.5, //, 5X, 21HLEFT WALL HEIGHT
    2 = ,F10.5,//,5X,21HRIGHT WALL HEIGHT = ,F10.5, //
    35x,12H MINIMUM = ,F10.5,//5x,6HDS = ,F10.5
    4,//5X,55HMEAN HEIGHT (ANNULAR REG.) - MEAN HEIGHT (CENTRAL REG.) =,
    5F10.5)
    CALL QUIK3L (-1, XL, XR, YB, YT, 42, BCDX, BCDY, JU, XX, YY)
     CALL QUIK3L( 0,1.0,0LIM,YB,YT,42,BCDX,BCDY,IK,X,Y)
 488 CONTINUE
```

STOP

ENO

CROSS-REFERENCE BETWEEN SYMBOLS (INPUT AND OUTPUT) SUBROUTINE FRED

Algebraic Symbol	FORTRAN Symbol	Description
a	Α	Radius of tank
α	ALPHA	Local angle of inclination of surface
Bo	во	Bond number
B _o H _o	воно	
$\sin \alpha_1$	SALP1	
$\sin \alpha_2$	SALP2	
$\cos \alpha_1$	CALP1	
$\cos \alpha_2$	CALP2	
$\frac{\mathrm{d}\alpha}{\mathrm{d}s} _{\mathrm{s}} = 0$	DADSO	Curvature on axis of symmetry
da ds	DALPDS	Curvature at general point
н _о	НО	Value of H on axis of symmetry
h_{m}	HM	Mean height of surface
$\frac{\Delta s}{2}$	HALFDS	Half of increment of arc length
$\Delta_{ m S}$	DS	Increment of arc length
	KOUNT	Number of iterations
	LL	Integer determining what data is printed (every 10 or every 10^2)
θ	THETA	Contact angle in radians

Algebraic Symbol	FORTRAN Symbol	Description
$\theta - \frac{\pi}{2}$	THET	
$\int_{0}^{s} \cos \alpha ds$	TCOS	$t\left(=\frac{r}{a}\right)$
$\int_0^{s} \sin \alpha ds$	TSIN	H - H _o
	XALP	α at wall
	YI	Storage location for surface heights
	XI	Storage location for radial distances

```
SUBROUTINE FRED(BO, THETA, DS, X, Y, II)
C*
C*
C*
       CALCULATE THE APPROXIMATIONS OF AXIALLY SYMMETRIC
C***
         STATIC SURFACES AT ZERO CONTACT ANGLE
C***
C*
C*
C*
      DIMENSION ALPHA(2 )
      DIMENSION X(2000), Y(2000), Z(10), W(10)
C
    CALL CAMRAV(935)
C
C
      HM=0.0
      A=1.0
      PI=3.1415927
    4 CONTINUE
    1 CONTINUE
  256 FORMAT(12+2F10.0)
      IF(L.EQ.1)G0 TO 9999
      IF(DS.GT.0.00005)LL=100
      IF(DS.GT.0.0005)LL=10
    9 FORMAT(1H ,2X,7H RH0= ,F10.5,/,3X,7H G= ,F10.5,/,
     13X,7H T= ,F10.5,/,3X,7H A= ,F10.5,/,3X,7H HM= ,F10.5,/,
```

41

```
23x,7HTHETA= ,F10.5,/,
    23X+7H US= +F10+5)
    KOUNT = U
    DX=.5
   5 FORMAT(5F1U.U)
1011 FORMAT(1H ,10X,3HB0=,1X,E16.8)
     WRITE(6,1011)80
     WRITE (6, 1012) THETA
1012 FORMAT(1H , 10X, 6HTHETA=, 1X, E16.8, 4H DEG )
     WRITE(6,1013)DS
1013 FORMAT(1H ,10X, 3HDS=,1X, E15.8)
    HALFUS=DS/2.
    THET=THETA-P1/2.
     DALPUS =.5
     DADSU=UALPDS
  10 H0=2.*DALPUS /80
     B0H0=B0*H0
     KOUNT = KOUNT+1
     IF (KOUNT.GT.100) GO TO 30
     TCUS=0.0
     TSIN=0.0
     X(1) = 0.0
    Y(1) = 0.0
     DALP=0.0
     IK=1
```

II=1

```
SALP1=0.0
```

CALP1=1.0

6 FORMAT(1H0,6HDALPD5,2X,F10.5,I4)

ALPHA(1)=0.0

20 CONTINUE

ALPHA(2)=ALPHA(1)+DALPUS*D5

SALP2=SIN(ALPHA(2))

TSIN= TSIN+(SALP2+SALP1)*HALFDS

SAVE=TCOS

CALP2=COS(ALPHA(2))

TCUS=TCOS+(CALP2+CALP1)*HALFDS

SALP1=SALP2

CALP1=CALP2

IJ=MOD(IK,LL)

IF(IJ.NE.0)GU TO 80

II=1I+1

X(II)=ICOS

Y(11)=TS1N

80 IK=IK+1

DALPDS=B0H0+B0*TSIN-SALP2/TCOS

ALPHA(1) = ALPHA(2)

IF (SAVE.GT. TCOS. AND. TCOS+.000001.LT.1.)60 TO 22

IF (TCOS+.000001.LT.1.0) GO TO 20

8 FORMAT(1H0+6HINTCOS+F10.5)

```
XALP=ALPHA(2)*57.29578
```

WEITE (0,060) XALP

7 FORMAT(1H0,6HALPHA ,F10.5)

IF (AbS(SAVE1-ALPHA(2)).LT..000001) GO TO 30

SAVE1 = ALPHA(2)

IF (AUS(ALPHA(2)+THET).LI...U002)GU TO 30

IF (ALPHA(2).LT. (U.B-THET)) GO TO 23

22 CONTINUE

DX=DX/2.

DAUSU=UADSO-UX

DALPUS=DADS0

GO TO 10

23 DX=DX/2.

DADSU=UADSU+UX

DALPOS=DADSU

GO TU 10

30 CONTINUE

II=II+1

X(II)=TCOS

Y(11)=151N

D0 660 I=1.II

 $Y(I) = HM/A + 2 \cdot /B0 * (DAUSO-COS(THETA)) + Y(I)$

666 CONTINUE

```
65 FORMAT (1H0+//+5x+5E15.8+/+5X+5E15.8)
```

660 FORMAT(1H0,10X,6HALPHA=,1X,E16,8,4H DEG)

WRITE(6,66)(X(J),J=1,II)

WRITE(0,66)(Y(J),J=1,II)

- 66 FORMAT(1H0,10F10.5)
- 67 FORMAT(1H1,5X,3HIK=,18)

RETURN

9999 CONTINUE

STOP

ENU

SUBROUTINE BC (A+B) A = B RETURN END

SUBROUTINE RPRINT(BCD,X)

WRITE(6,1401)8CD,X

1401 FORMAT (1H +12H*** TEST ***+10X+A6+1X+2H= E15+8)

RETURN

ENU

SAMPLE DATA

5.0	2.	3.	
10.			
20.			
50.			
75.			
100.			
250.			
500.			

DOCUMENT CONTROL DATA - R&D (Security classification of title, body of abatract and indexing amount to antared when the overall report in classified)			
1. ORIGINATIN & ACTIVITY (Corporate author)		2. REPOR	T SECURITY CLASSIFICATION
Research Laboratories, Brown Engineering		None	
Company, Inc., Huntsville, Alabama		2.6. GROUP	None
3. REPORT TITLE		£ <u>,,.,.,</u> ,	
Liquid-Gas Interface Configurations in	n Interconnecte	d Conc	entric Cylinder
Tankage Systems			
4. DESCRIPTIVE NOTES (Type of report and inclusive dates)			
Technical Note R-244			
5. AUTHOR(S) (Last name, first name, initial)	······································		
Geiger, Dr. Frederick W. and May,	James C.		
6. REPORT DATE September 1967	74. TOTAL NO. OF P ~ 49	AGES	75. NO. OF REFS 5
S. CONTRACT OR GRANT NO.	94. ORIGINATOR'S RE	PORT NUM	5 E R(S)
NAS8-20073			
6. PROJECT NO.	Technical N	lote R-	244
с.	9b. OTHER REPORT this report)	NO(S) (Any	other numbers that may be essigned
d.	Non	e	
10. A VAIL ABILITY/LIMITATION NOTICES			
None			
11. SUPPLEMENTARY NOTES	12. SPONSORING MILI	TARY ACTI	VITY
None	Propulsion a	and Veh	icle Engineering Lab.
	George C. N Huntsville.	/arshal Alabam	l Space Flight Center a
13. ABSTRACT			14. KEY WORDS
A computer program has been deve and analytically define the static shap	eloped to graph e of the liquid-	ically gas	hydrostatics
interface in the annular region betwee	n any two conc	entric	fluid mechanics
cylinders in an axial force field for an contact angle. The program also incl	udes as a subr	r and outine	free surface
a previously reported method of calcu	lating the stati	ic	low g
shape of the liquid-vapor interface within a single cylindri			Bond number
annular and central regions are given for Bond numbers			surface tension
(with inner cylinder radius as characteristic length)			
ranging from 10 to 500, for a contact angle of 5 degrees,			
and for a radius ratio of 1.5.			
		1	
I a second s			