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STUDY OF A NONSTATIONARY ADJOINT  
STATISTICAL WIND MODEL FOR FLIGHT  
CONTROL SYSTEMS ANALYSIS

ENGINEERING REPORT NO. 1621  
28 FEBRUARY 1969

PREPARED FOR  
GEORGE C. MARSHALL SPACE FLIGHT CENTER  
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
HUNTSVILLE, ALABAMA

FINAL REPORT  
CONTRACT NO. NAS8-21444

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CONTROL SYSTEMS ANALYSIS

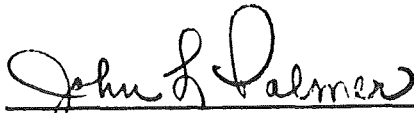
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Prepared by:



J. L. Palmer  
Project Manager

Approved by:



C. M. Askey  
Program Manager

## SUMMARY

Design and analysis of large booster flight control systems require techniques that account for the time varying character of the dynamic system. In addition, the techniques must account for system forcing functions that are random processes. The tools available for these studies have been limited. In general, they are subject to analytical formulation, but tend toward numerical schemes. Such a numerical scheme was presented in "Launch Vehicle Winds and Turbulence Response by Nonstationary Statistical Methods," NASA CR846.

This report is an extension of the work undertaken in NASA CR846 and presents the application of techniques there formulated, to large amounts of Jimsphere data with the random process frequency content included.

The work was performed at Hayes International Corporation from June 1968 to February 1969 under Contract No. NAS8-21444. "Study of a Nonstationary Adjoint Statistical Wind Model for Flight Control Systems Analysis," for Aero Astrodynamics Laboratory of George C. Marshall Space Flight Center.

Mr. Jerome R. Redus and Mr. William W. Heuser of the Optimal Control Theory Branch of the Astrodynamics and Guidance Theory Division were the NASA technical supervisors during the periods of performance of this contract.

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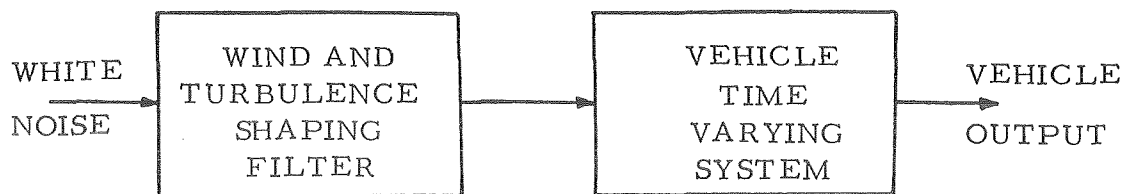
## I. INTRODUCTION

For time varying systems that ascend the atmosphere, winds are a major consideration in design of flight control systems. Therefore, it becomes advantageous to have available a tool for design and analysis that can rapidly produce accurate vehicle response statistics. This tool is available in the winds and turbulence "shaping filter" and the "adjoint" techniques of analysis.

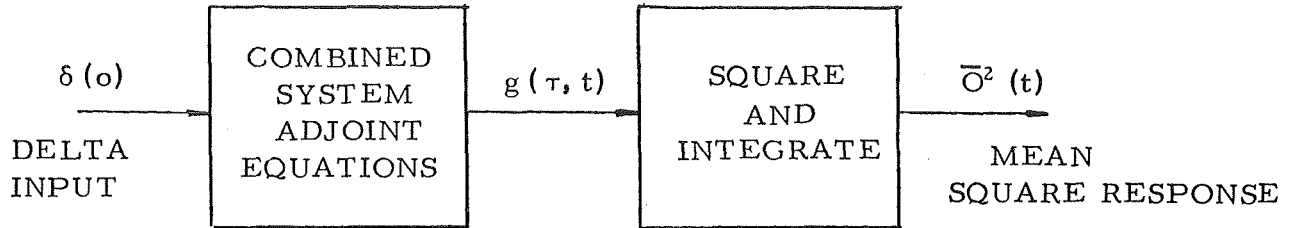
The shaping filter is designed for use with the adjoint technique of analysis, or programs formulated with optimal control techniques requiring mean square system response. By determining the appropriate coefficients, the shaping filter can be used to generate the winds and turbulence statistics that are necessary as an input to a dynamic system to obtain the system mean square response due to winds and turbulence input.

The shaping filter developed in this report is representative of the winds and turbulence over Cape Kennedy, Florida as measured by the Jimsphere Wind Sounding Program conducted by Marshall Space Flight Center, Huntsville, Alabama.

To use the shaping filter with an adjoint technique of analysis, the shaping filter is joined with the time varying system to form a combined system, that has a white noise input and a response which is the vehicle output that normally results from winds and turbulence.



The combined system equations are programmed as adjoint equations.



The impulse response of the adjoint system is the adjoint weighting function of the system,  $g(\tau, t)$ , as a function of the second argument  $\tau$ . Application of linear system theory has resulted in the equation

$$\bar{\sigma}^2(t) = \int_0^t g(\tau, t)^2 d\tau,$$

and allows direct calculation of the same time varying system mean square response that would result from an ensemble of wind and turbulence profiles input to the time varying system.

The shaping filter, so named because it shapes a white noise input into a prescribed random process whose correlation function is known, was suggested for use with the adjoint technique of analysis by Lanning and Battin <sup>(1)</sup>, for stationary processes, and applied to nonstationary processes by Bailey, Palmer, and Wheeler <sup>(2)</sup>, using statistical dynamics based on the works of Batkov <sup>(3)</sup>, and Solodovnikov <sup>(4)</sup>.

The general synthesis of the shaping filter in the nonstationary case requires far more preparation than synthesizing the shaping filter for a stationary process developed in Ref. (1). This is primarily due to the fact that the power spectra and correlation of stationary processes have a direct transfer relationship and admit to an analytical solution. There are relationships in the nonstationary case, but at present there are no analytical solutions which allow the attainment of the filter equation through direct transformation as in the stationary case.

The shaping filter is defined as an  $n^{\text{th}}$  order differential equation with time or altitude varying coefficients that are developed from the winds and turbulence data. The input to the shaping filter is white noise and the output winds and turbulence statistics equivalent to ensemble statistics of the winds and turbulence used to develop the filter. These ensemble winds and turbulence nonstationary statistics, correlation or covariance, are related to the filter characteristics through the filter weighting function.

The winds and turbulence data are reduced and arranged in a form such that the statistics may be related to the filter. The order of the filter must be established and the necessary derivatives of the covariance determined. These derivatives are used to determine the coefficients of the filter equation.

Determination of the time varying coefficients of the shaping filter is equivalent to development of the shaping filter.

Section I of this report presents the basic equations that have been developed as foundation stones for the nonstationary statistical development. A

complete development of these equations is given in References 3 and 4, with an excellent summary of the adjoint approach in Reference 5. Sections II, II and IV present the statistical development and underlying problems that are to be considered prior to undertaking the numerical task of development of the mean and covariance matrix.

Section V presents the approach taken in developing the derivatives of the covariance at each point along the altitude span of the covariance matrix.

Section VI presents development of the coefficients for both the left and right hand side of the equation (1) of Section II.

Section VII presents the filter coefficient development computer programs.



## II. BASIC RELATIONSHIPS

To arrive at the basic equations that are to be used in the numerical and analytical presentation and development of the shaping filter, let the filter be defined as an  $n^{\text{th}}$  order differential equation with a white noise input

$$a_n(t) \frac{d^n x(t)}{dt^n} + \dots + a_0(t)x(t) = b_m \frac{d^m \eta(t)}{dt^m} + \dots + b_0(t)\eta(t)$$

or symbolically

$$D(p, t)x(t) = M(p, t)\eta(t) \quad \begin{aligned} D(p, t) &= \sum a_i(t) \frac{d}{dt} \\ M(p, t) &= \sum b_i(t) \frac{d}{dt} \end{aligned} \quad p = \frac{d}{dt} \quad (1)$$

$$m < n, \quad \eta(t) = \text{White Noise}$$

The solution to the shaping filter problem is determination of the  $a_i(t)$ ,  $b_i(t)$  of equation (1) such that  $x(t)$  has prescribed nonstationary statistics in the form of an arbitrary continuous covariance. A solution to the filter equation may be written as

$$X(t) = \int_{-\infty}^t g(t, \tau) \eta(\tau) d\tau. \quad (2)$$

$X(t)$  is the random process desired and  $g(t, \tau)$  is the filter weighting function, or impulse response. The impulse response  $g(t, \tau)$  of the filter can be defined as the solution to

$$D(p, t)g(t, \tau) = M(p, t)\delta(t - \tau). \quad (3)$$

Equation (3) plays a major role in the development of relationships between the filter equation and the winds and turbulence, that the filter is to generate, for it delineates the form of the equations to be developed.

The form that is desired is a form that allows the solution to the filter equation to be in terms of winds and turbulence statistics, namely, correlation or covariance. The correlation of the solution to equation (1) is the average product of equation (2) at times  $t$  and  $\tau$ .

$$\begin{aligned}
 R_x(t, \tau) = \overline{x(t)x(\tau)} &= \int_{-\infty}^t g(t, \lambda) \eta(\lambda) d\lambda \int_{-\infty}^{\tau} g(\tau, \theta) \eta(\theta) d(\theta) \\
 &= \int_{-\infty}^t \int_{-\infty}^{\tau} g(t, \lambda) g(\tau, \theta) R_{\eta}(\theta, \lambda) d\lambda d\theta
 \end{aligned}
 \tag{4}$$

where  $R_{\eta}(\theta, \lambda)$  is the correlation of the input which is known to be  $\delta(t)$ , a result following from the input being white noise.

Substituting  $\delta(t)$  into equation (4) and recalling that

$$\int_0^t g(t, \tau) \delta(\tau - \lambda) dx = g(t, \lambda)
 \tag{5}$$

equations for  $R_x(t, \tau)$  are derived

$$\begin{aligned}
 R_x(t, \tau) &= \int_{-\infty}^t g(t, \lambda) g(\tau, \lambda) d\lambda \text{ for } t < \tau \\
 &= \int_{-\infty}^{\tau} g(t, \lambda) g(\tau, \lambda) d\lambda \text{ for } t > \tau
 \end{aligned}
 \tag{6}$$

Operating on both sides of equation (6) with  $D(p, t)$ , equation (6), part two reduces to zero for  $\tau < t$ . The correlation, or covariance, of the output of the filter satisfies part two of equation (7), hence,

$$\begin{aligned}
 D(p, \tau) R_x(t, \tau) &= 0 && t < \tau \\
 D(p, \tau) R_x(t, \tau) &= M(p, \tau) g(t, \tau); && t > \tau; \quad p = \frac{d}{d\tau}
 \end{aligned}
 \tag{7}$$

Equations (7) state the relationships between the correlation of the winds and turbulence, and the filter function. They are the basic equations from which the  $a_i(t)$ , the time varying coefficients of the left-hand side of equation (1) are developed. The values of equation (1) on the right-hand side,  $b_i(t)$ , are developed from the initial conditions placed on  $g(t, \tau)$ . The initial conditions used to develop the  $b_i(t)$  have been thoroughly developed in Reference 2.

The numerical techniques that follow are developed in terms of altitudes  $h_i$  and  $h_j$  as opposed to  $t$  and  $\tau$  in the development presented in Section II. Transformation from the altitude domain to the time domain can be accomplished by techniques described in Reference (6).

## III. WINDS AND TURBULENCE DATA DEVELOPMENT

The shaping filter must have a random output which has the same correlation or covariance as the winds and turbulence measured by FPS/16-Jimsphere techniques. Figure 1 represents a Jimsphere zonal wind speed recording plotted over the altitude interval of interest.

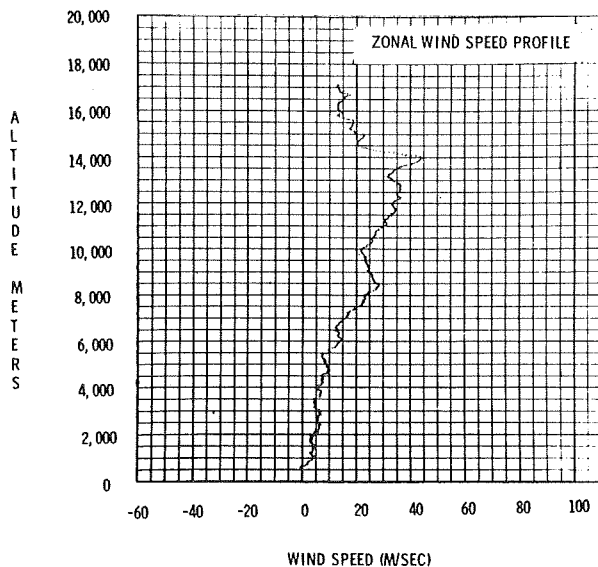


Figure 1

From an ensemble of such profiles, the mean and covariance matrices are to be formed. Utilization of the Jimsphere data to develop a shaping filter requires a careful evaluation of both Jimsphere data and characteristics of the vehicle to be analyzed. The Jimsphere data, at present, is represented by over 1200 records of Cape Kennedy winds, recorded at 25 meter intervals, beginning at approximately 200 meters and extending to approximately 18000 meters. Present records are zonal, meridional and scalar winds.

Such large quantities of data require selectivity in use such that the model contains statistics that excite the important vehicle modes that would be excited

during an actual flight. Selecting data for building the filter can best be explained by an example. Consider the Saturn class vehicle during actual flight. The rigid body mode is in the neighborhood of 1 radian per second and the first the three elastic modes are approximately 6, 13, and 18 radians per second. The frequency range of importance then will be between approximately 0.5 and 20 radians per second.

The model can be developed around statistics that contain these frequencies if the proper lag is utilized in development of the covariance matrix. In order to determine a lag size in the covariance matrix that will insure incorporating frequencies up to 20 rad/sec into the problem, consider a vehicle velocity of 525 m/sec, and wind data recorded at 25 meter intervals. Then,

$$\begin{aligned} \lambda(\text{wind profile wavelength}) &= \frac{V (\text{vehicle velocity at max } Q)}{f (\text{vehicle frequency of interest})}. \\ &= \frac{525 \text{ m/sec}}{3.18 \text{ c/s}} = 165 \text{ meters} \end{aligned} \tag{8}$$

Therefore, a wavelength of 165 meters is required to introduce frequencies of up to 20 rad/sec into the vehicle problem in the maximum dynamic pressure region. When building a covariance matrix, three points are required to define one wavelength. Hence, the lag size must be one-half (1/2) the wavelength requirement to avoid averaging out the desired frequencies. Therefore, to use the Jimsphere data without interpolation, a lag size of 75 meters would be appropriate in the example considered.

By considering the frequency range of interest, the analyst can select from the bulk of Jimsphere data only the wind direction desired, at 75 meter intervals,

thereby reducing the bulk of data to be handled by from 3 to 5 times. Proper selection of data can be crucial where the filter is to be developed on a small digital computer, i. e., 5K storage.

The data selected for this model development were the EAST-WEST (zonal) component of all Jimsphere recordings available. The data were taken in 50 meter steps, beginning at 200 meters and extending through 16,050 meters. To facilitate the analysis, all E-W (zonal) data at 50 meter intervals on 5 Jimsphere record tapes were transferred to a single E-W tape at 50 meter intervals, reducing the data handling from 5 tapes to 1 tape.

The single tape contains record lengths of 1251 words: altitude, and 1250 recordings at a given altitude, extending over several years, for a grand total of 320 records each of 1251 words. From the ensemble of such profiles, on the single tape, separate statistics may be readily calculated for any altitude range. The required statistics of interest are:

mean wind at altitude  $h_r$

$$V_w(h_r) = \frac{1}{N} \sum_{j=1}^N V_w(h_r)_j / N, \quad (9)$$

covariance of the wind profile

$$C(V_w(h_r), V_w(h_s)) = \frac{1}{N} \sum_{j=1}^N (V_w(h_r) - \bar{V}_w(h_r))_j (V_w(h_s) - \bar{V}_w(h_s))_j / N, \quad (10)$$

and correlation defined as

$$R(V_w(h_r), V_w(h_s)) = C(V_w(h_r), V_w(h_s)) / (V_w(h_r) V_w(h_s)) \quad (11)$$

when the average of the process is equal to zero.

Either covariance or correlation may be used in the filter development. However, previous developments have indicated that covariance is a more tractable number and will be used in the filter development.

Utilizing equations (1) and (3), and the reduced data tape, a covariance and mean matrix can be formed.

## IV DEVELOPING THE COVARIANCE MATRIX

Referring to Figure 2, a typical set of N wind profiles, and applying standard ensemble techniques, a matrix of covariance values may be formed.

A sample average or expected value of the wind velocity at any altitude  $h_r$  may be calculated by use of equation (9).

$$AV(Vh_r) = E(V(h_r)) = \frac{1}{N} \sum_{j=1}^N V_j(h_r) \quad (12)$$

$\frac{1}{N} \sum_{j=1}^N V_j(h_r)$  may also be expressed  $\langle V_j(h_r) \rangle$ , the ensemble average of wind velocity at altitude  $h_r$ .

Let  $V'(h) = V(h) - \langle V_j(h) \rangle$ . Then the covariance is defined as

$$\begin{aligned} \text{Cov}(h_r, h_s) &= \langle V'(h_r) V'(h_s) \rangle = \{ V(h_r) - \langle V_j(h_r) \rangle \} \{ V(h_s) - \langle V_j(h_s) \rangle \} \\ &= \langle V(h_r) V(h_s) \rangle - \langle V_j(h_r) \rangle \langle V_j(h_s) \rangle \end{aligned} \quad (13)$$

Equation (13) indicates the covariance value between wind velocity at altitudes  $h_r$  and  $h_s$  is given by the correlation of wind velocities at altitudes  $h_r$  and  $h_s$  averaged across the wind profile sample, minus the product of the average values of wind velocity at altitudes  $h_r$  and  $h_s$ . By taking altitudes  $h_r$  and  $h_s$  throughout the wind profile, from the lowest to the highest altitude, a matrix of covariance values will be formed. The matrix is shown in Figure 3, and is an  $n \times n$  symmetric matrix with  $n$  depending on the lag size selected.

The numerical scheme that is used with equations (12) and (13) depends on the computational facility available, and as more Jimsphere data becomes



available, the computational scheme will play an even more important part. The scheme used in computing the matrix of this program was to "split out" the 320 Jimsphere records into 4 tapes of 80 records each, thus saving computer time in search through the tape as the record number increases.

There were 1251 words read into computer storage for 2 altitudes, beginning with 200 meters altitude. These records were correlated with themselves and all records above 200 meters. This procedure is repeated throughout the 320 profiles. Any number of records may be read in initially, depending on available storage space of the computational facility. The technique described required 92 minutes on an IBM 7094 computer. An IBM 360-30 of 5K storage was tried but proved prohibitive timewise. This machine appeared to be execute bound.

The covariance matrix, generated and stored rowwise, is now ready for calculating the  $n$  derivatives for use with regression analysis to develop the coefficients.

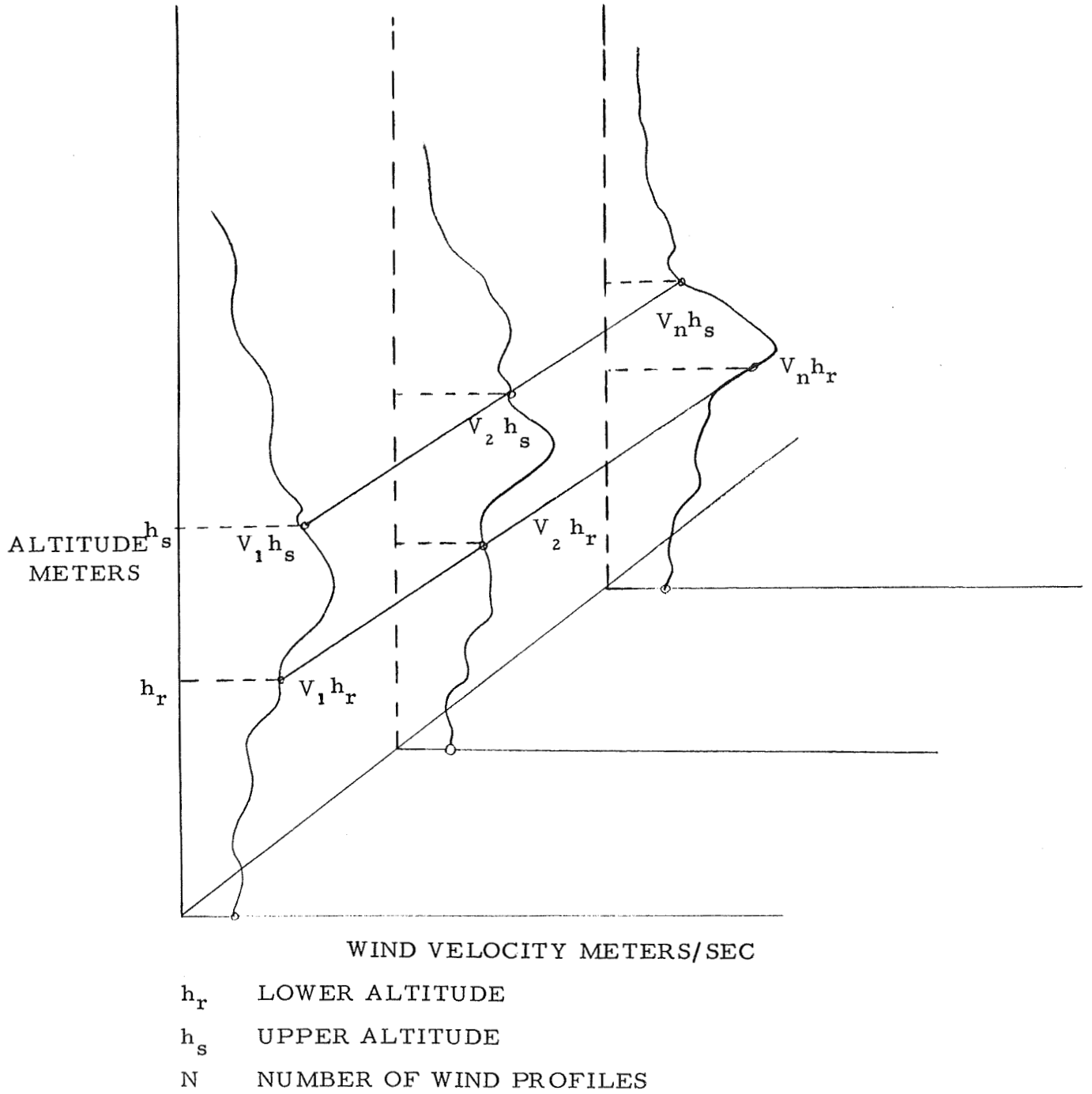
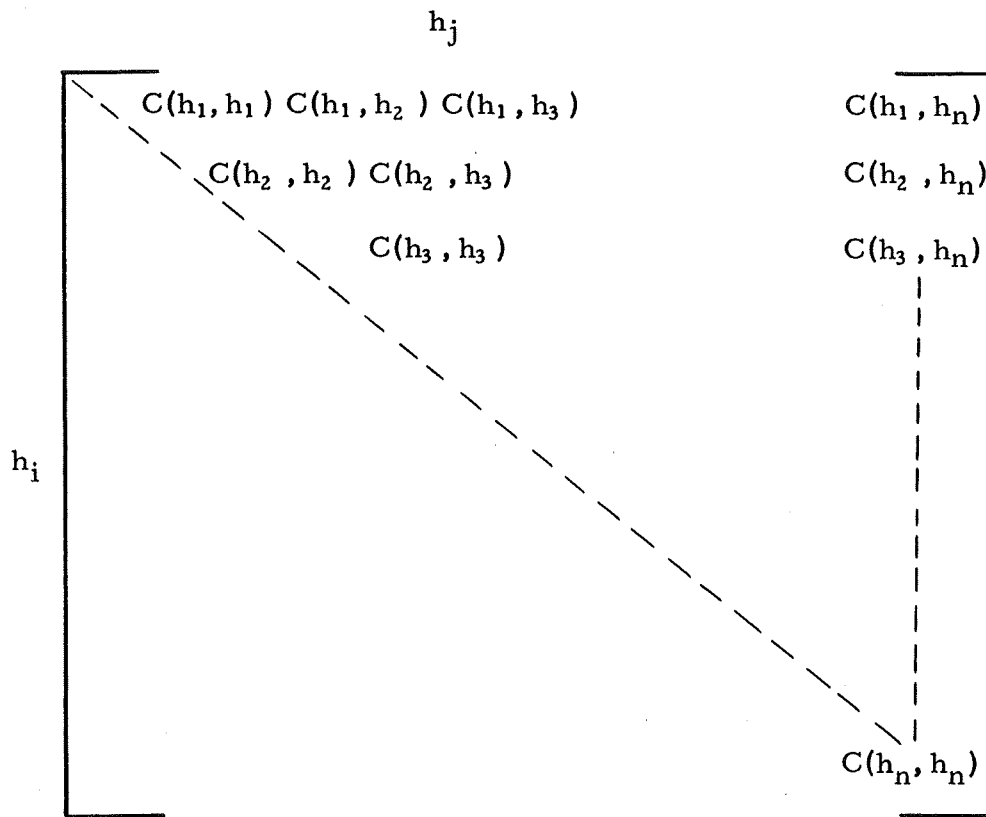


Figure 2

COVARIANCE MATRIX



$N \times N$  Symmetric Matrix

Figure 3

V. DETERMINATION OF THE DERIVATIVES  
OF THE COVARIANCE MATRIX

A method for developing  $\frac{d}{dh_j} C(h_i, h_j)$  and  $\frac{d^2}{dh_j^2} C(h_i, h_j), \dots, \frac{d^n}{dh_j^n} C(h_i, h_j)$

when  $C(h_i, h_j)$  is given in tabulated form as shown in Figure 3 is to represent  $C(h_i, h_j)$ , as shown in Figure 4, as a polynomial and evaluate  $\frac{dC}{dh}$  and  $\frac{d^2C}{dh^2}, \dots, \frac{d^n C}{dn^n}$  along the span of the polynomial. The polynomial that represents  $C(h_i, h_j)$  is called an interpolating polynomial and may be found from the tabulated data through the use of a finite difference table to develop the elements of the polynomial.

When tabulated data is given as in the case of  $C(h_i, h_j)$  of Figure 3,  $h_i$  is held constant for any tabulation of  $C(h_i, h_j)$  resulting in a  $C(h_i, h_j)$  that has  $h_i$  as a dependent variable and  $h_j$  as an independent variable.

The elements of the finite difference table are developed from the tabulated  $C(h_j)$ , called the zero order difference. Representing the first difference in the finite difference table as  $\Delta C_0 = C(h_{j+1}) - C(h_j)$ , the difference in  $C(h_j)$  over a span of altitude  $h$  (lag size of  $C$ ),  $\Delta^2 C_0$  as the second difference etc., the table takes the form shown in equation (14). The whole curve made up of all covariance values for a fixed  $h_i = 200$  meters is shown in Figure 4. Determination of the values of  $\frac{dC}{dh_j}$  by fitting this curve with a single polynomial would require a polynomial of prohibitive degree if an accuracy of interpolation of 1% is desired. This problem was circumvented by using a floating point polynomial fit of small segments of the total curve, in this case 2000 meters and a fourth degree polynomial.

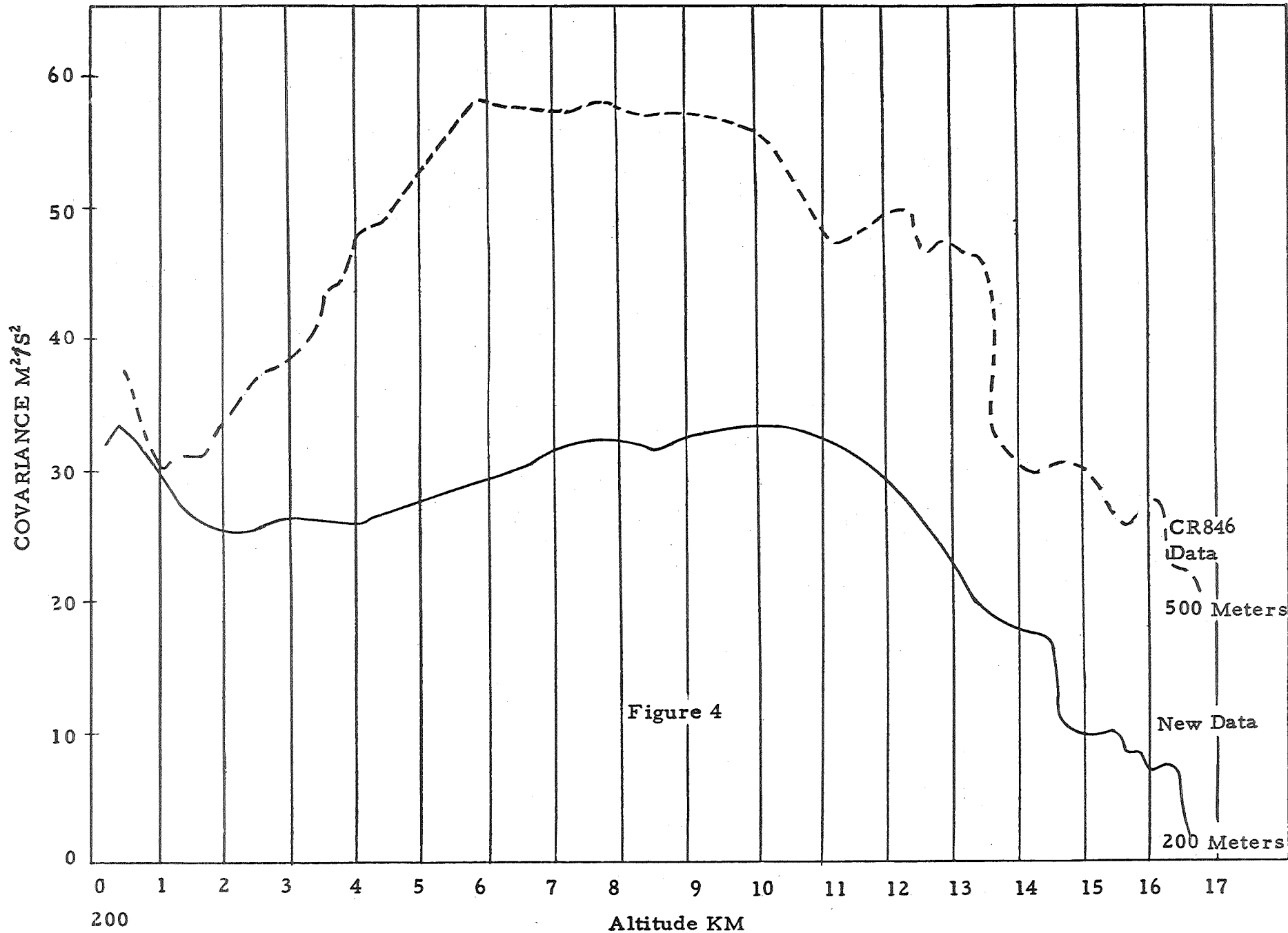


Figure 4

$C_0(h_0)$					
	$\Delta C_0$				
$C_1(h_1)$		$\Delta^2 C_0$			
	$\Delta C_1$				
$C_2(h_2)$			$\Delta^2 C_1$		
	$\Delta C_2$				
$C_3(h_3)$				$\Delta^n C_0$	
'					
'					
'					
$C(h_n)$					

(14)

From this finite difference table, the interpolation of C may be accomplished. The Gregory-Newton formula for forward interpolation will be used. This equation is

$$P_n(C) = \sum_{r=0}^n \frac{u^{[r]}}{r!} \Delta^r C_0 \tag{15}$$

$$u^{[r]} = u(u-1)(u-2) \dots (u-r+1)$$

The polynomial is some function of  $C(h_j)$ ,  $f(C)$ . It is approximated by the polynomial  $P_n(u)$  where  $u = \frac{C_1 - C_0}{h}$ ,  $h = .05$  kilometer.

$$\frac{df(C)}{dh_j} = \left( \frac{dP_n(u)}{du} \right) \frac{du}{dh_j} \quad (16)$$

$$\frac{du}{dh_j} = \frac{1}{h}$$

Differentiation of  $P_n(u)$  results in derivative matrices similar to the covariance matrix. The derivative matrices have as elements the first and second, or more, derivatives of covariance corresponding to the elements of the covariance matrix. The  $N$  derivative matrices are developed from the covariance matrix by using the row-wise differentiation of the covariance matrix. The elements of the covariance, first, second, third and etc., derivative matrices are used with the regression technique of Section VI to develop the  $a_i(t)$  of equation (1).

As in the initial selection and organization of the winds and turbulence data, the matrices must be arranged in a convenient form if a small computer is to be used for the filter development.

The elements of all matrices are formed by row-wise operation whereas the regression analysis of Section VI, used for developing the  $a_i(t)$ , is performed column-wise. To facilitate the regression analysis, data sorting is performed on the three matrices to form a single matrix. The single matrix is unique in that each single element of this new matrix has several parts,  $C_{ij}, \frac{dC_{ij}}{dh_j}, \dots, \frac{d^n C_{ij}}{dh_j^n}$ . Therefore, when the regression analysis is performed at a given  $h_j$ , this matrix furnishes all necessary data values in a single record read by the computer. Regression analysis begins with the second record, since the

first record contains only one point i. e.  $(\text{Cov}(200, 200), \frac{dC}{dn}(\text{Cov}(200, 200), \frac{d^n}{dh^n}(\text{Cov}(200, 200))))$ , and the regression analysis does not hold (is equal to zero) for a single point.



VI. REGRESSION ANALYSIS FOR THE SHAPING  
FILTER EQUATION COEFFICIENTS

In general theory, multivariate regression plane is represented by

$$X_{1r} = k_1 + k_2 X_2 + k_3 X_3 \dots k_n X_n \quad (17)$$

where  $X_1$  is a regression on  $X_2 \dots X_n$ . In other words,  $X_1$  is the dependent variable and it remains to determine the  $k_i$  such that the equation best approximates the data represented by the  $X$  variables. By relating the variables  $C(h_i, h_j)$ ,  $C^1(h_i, h_j)$ ,  $\dots$ ,  $C^n(h_i, h_j)$  of the problem to the regression equation, the  $a_i(t)$  of equation (1) can be found as the  $k_i$  are found in the general problem. The filter order is selected as 2 for the problem filter. The regression equation and the filter equation as related to the covariance is  $C^n(h_i, h_j)$ .

$$C'' + a_1 C + a_0 C = 0 \quad (18)$$

This equation is similar to the regression equation

$$X_{1r} = k_2 X_2 + k_3 X_3 \quad (19)$$

where the  $X_i$  are referred to their mean value and  $k_1$  becomes zero. The  $a_i(h)$  are to be found such that the sum of the squares of the deviations between the observed and predicted values, as given by

$$D^2 = \sum_{j=1}^N \{C''_j + a_1 C_j + a_0 C_j\}^2, \quad (20)$$

is a minimum. By taking the partial derivative of  $D^2$ , a set of normal equations are arrived at such that their solution results in the  $a_i$  given in terms of products and squares of known values of  $C$ ,  $C^1$ ,  $C''$  that were stored in the single matrix generated in the final stages of the derivative program. The equations

that result from differentiation of equation (20) may be solved as to equations with two unknowns, resulting in expressions for  $a_0(h_i)$ ,  $a_1(h_i)$ ,  $i = 200, 15950$  meters, by 50 meter steps.

$$a_0(h_i) = \frac{-\sum_{j=1}^i C''(h_i, h_j) \sum_{j=1}^i C(h_i, h_j)^2 + \sum_{j=1}^i C'(h_i, h_j) C''(h_i, h_j) \sum_{j=1}^i C(h_i, h_j) C'(h_i, h_j)}{\sum_{j=1}^i C(h_i, h_j)^2 \sum_{j=1}^i C'(h_i, h_j)^2 - (\sum_{j=1}^i C(h_i, h_j) C'(h_i, h_j))^2} \tag{21}$$

$$a_1(h_i) = \frac{-\sum_{j=1}^i C(h_i, h_j)^2 \sum_{j=1}^i C'(h_i, h_j) C''(h_i, h_j) + \sum_{j=1}^i C(h_i, h_j) C'(h_i, h_j) \sum_{j=1}^i C(h_i, h_j) C'(h_i, h_j)}{\sum_{j=1}^i C(h_i, h_j)^2 \sum_{j=1}^i C'(h_i, h_j)^2 - (\sum_{j=1}^i C(h_i, h_j) C'(h_i, h_j))^2}$$

The numerical approach taken to develop the  $b_i(h)$  can be more easily understood by referring to Figure 5 during the following development. When the  $C(h_i, h_j)$  of Figure 3 is plotted with  $h_i$  as the dependent variable, the curves for  $h_i > h_j$  are obtained and when the  $C(h_i, h_j)$  are plotted with  $h_j$ , the dependent variable, the curves of  $h_i < h_j$  are obtained. They are identical due to the symmetry of the matrix. Either form may be used to generate the filter coefficients. For the case taken,  $h_i$  is the dependent variable and the curves for  $h_j > h_i$ , the curves to the right of  $h_i = h_j$  are used.

The line  $h_i = h_j$  is of importance in evaluating  $\left. \frac{dc}{dh_j} \right|_{h_i = h_j^-}$  and  $\left. \frac{dc}{dh_j} \right|_{h_i = h_j^+}$ .

$\left. \frac{dc}{dh_j} \right|_{h_i = h_j^+}$  is evaluated along the curves B at a point  $h_i = h_j^+$ . The value of  $\left. \frac{dc}{dh_j} \right|_{h_i = h_j^-}$  is evaluated along the curves A at a point  $h_i = h_j^-$ , or

can be more readily found by differentiating curves C since the values of  $C(h_i, h_j)$  etc. are taken along the curves B at 50 meter intervals. The values of  $C(h_i, h_j)$  at  $h_i = h_j$  appear to be zero (0) in Figure 5. This, however, is not the case as is shown in the exploded view of Figure 5, where a discontinuity is evident.

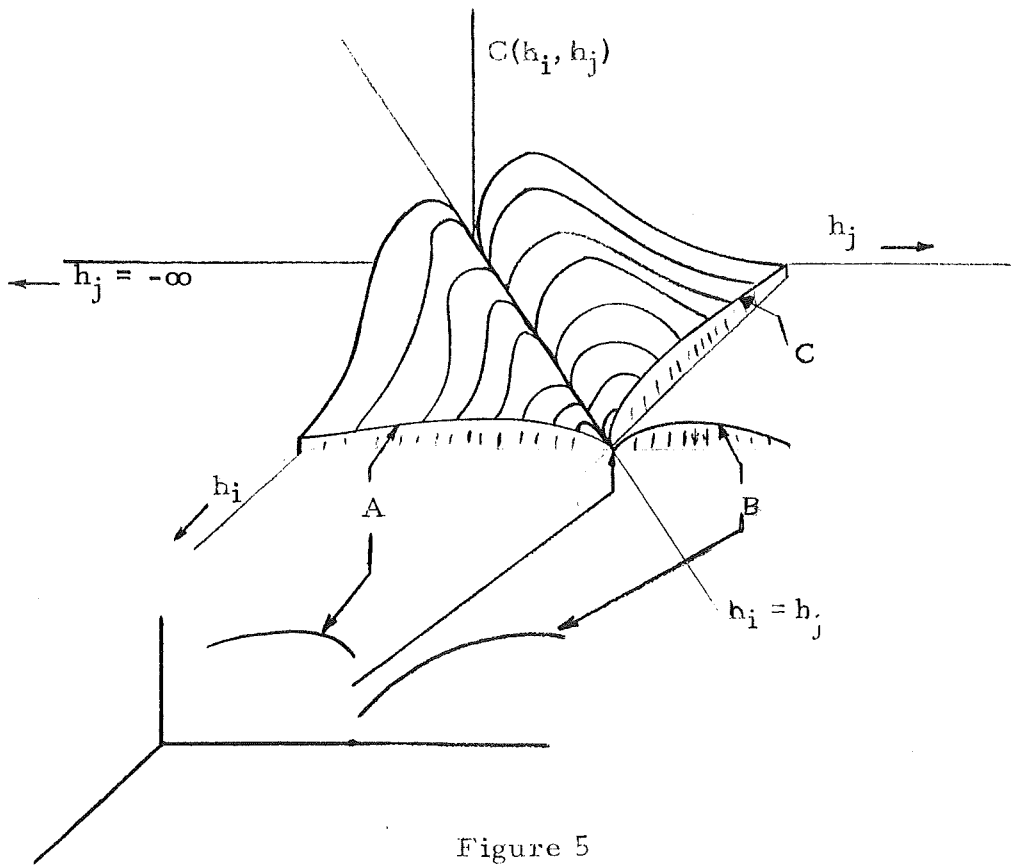


Figure 5

The curves A and C are obtained by taking a plane through the  $C(h_i, h_j)$  curves at points  $h_i$  or  $h_j$ .

Since  $g(h_i, h_j)$  is the impulse response function or weighing function for this filter

$$a_0(h_i) g(h_i, h_j) + a_1(h_i) \frac{dg(h_i, h_j)}{dh_j} + \frac{d^2 g(h_i, h_j)}{dh_i^2} = 0 \text{ for } h_i > h_j .$$

By approximating  $\frac{dg(h_i, h_j)}{dh_j}$  and  $\frac{d^2 g(h_i, h_j)}{dh_i^2}$  in terms of differences representing three points, a system of equations may be obtained to be utilized for numerically obtaining  $g(h_i, h_j)$  for points near  $h_i = h_j$ . If  $h_{i0}$ ,  $h_{i1}$  and  $h_{i2}$  are three consecutive values of  $h_i$  near  $h_i = h_j$  taken along the span of  $h_j$ , then

$$\left. \frac{d g(h_i, h_j)}{dh_j} \right|_{h_i = h_{i0}} = \frac{1}{2} [ -3 g(h_{i0}, h_j) + 4 g(h_{i1}, h_j) - g(h_{i2}, h_j) ]$$

$$\left. \frac{d g(h_i, h_j)}{dh_j} \right|_{h_i = h_{i1}} = \frac{1}{2} [ -g(h_{i0}, h_j) + g(h_{i2}, h_j) ]$$

$$\left. \frac{d g(h_i, h_j)}{dh_j} \right|_{h_i = h_{i2}} = \frac{1}{2} [ g(h_{i0}, h_j) - 4 g(h_{i1}, h_j) + 3 g(h_{i2}, h_j) ]$$

and

$$\left. \frac{d^2 g(h_i, h_j)}{dh_j^2} \right|_{h_i = h_{i0}} = g(h_{i0}, h_j) - 2 g(h_{i1}, h_j) + g(h_{i2}, h_j)$$

In terms of three consecutive points the second derivatives are assumed constant. The points are taken close enough together to minimize error.

From the preceding numerical substitutions for the first and second derivatives of  $g(h_i, h_j)$  in the differential equation of the shaping filter, the following system of difference equations may be obtained.

$$\begin{aligned}
 & [ a_0(h_{i1}) - 2 a_1(h_{i1}) ] g(h_{i1}, h_j) + a_1(h_{i1}) g(h_{i2}, h_j) = \\
 & \qquad [ a_0(h_{i0}) - 2 a_1(h_{i0}) + a_1(h_{i1}) ] g(h_{i0}, h_j) \\
 & ( a_0(h_{i1}) + 1 a_1(h_{i2}) ] g(h_{i1}, h_j) + ( a_1(h_{i1}) - a_0(h_{i2}) - 2 a_1(h_{i2}) - 2 a_1(h_{i2}) ] \\
 & \qquad g(h_{i2}, h_j) = a_1(h_{i1}) g(h_{i0}, h_j).
 \end{aligned}$$

as developed in reference (3).

$$g(h_{i0}, h_j) = \left( \left. \begin{array}{c} \frac{d C(h_{i0}, h_j)}{dh_{i0}} \\ \frac{d C(h_{i0}, h_j)}{dh_{i0}} \end{array} \right|_{\substack{h_{i0} = h_j \\ h_{i0} = h_j}} \right)$$

Thus, the system of two equations and two unknowns can be solved for  $g(h_{i1}, h_j)$  and  $g(h_{i2}, h_j)$ . Substituting  $g(h_{i0}, h_j)$ ,  $g(h_{i1}, h_j)$ ,  $g(h_{i2}, h_j)$  when

$$h_j = h_{i0} \text{ in the formula for } \left. \frac{d g(h_i, h_j)}{dh_j} \right|_{\substack{h_i = h_{i0} \\ h_j = h_{i0}}}$$

enables one to compute  $b_1(h_{i0})$  from the formulas,

$$b_1(h_{i0}) = g(h_i, h_j) \left. \begin{array}{c} \\ h_i = h_{i0} \\ h_j = h_{i0} \end{array} \right| \text{ and}$$

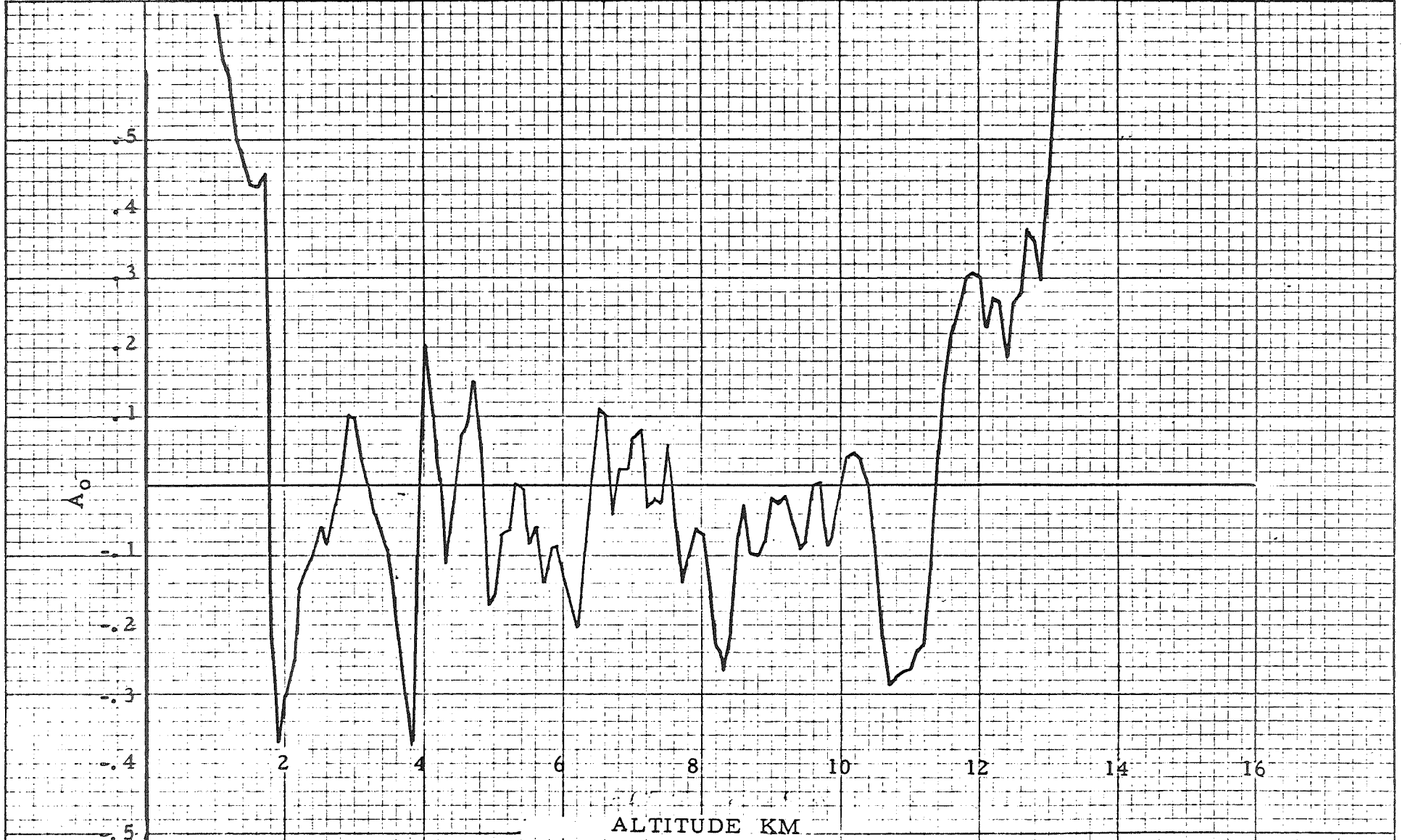
$$b_0(h_{i0}) = \frac{dg(h_i, h_j)}{dh_j} \left. \begin{array}{c} \\ h_i = h_{i0} \\ h_j = h_{i0} \end{array} \right| + a_1(h_{i0}) b_1(h_{i0}).$$

## VII. SHAPING FILTER COEFFICIENTS

The coefficients of the shaping filter differential equation (1) are given in this section. Figures 6 through 9 are  $A\phi$ ,  $A1$ ,  $B\phi$ ,  $B1$  respectively. Computer tabulations of each graph are given following each graphical display.

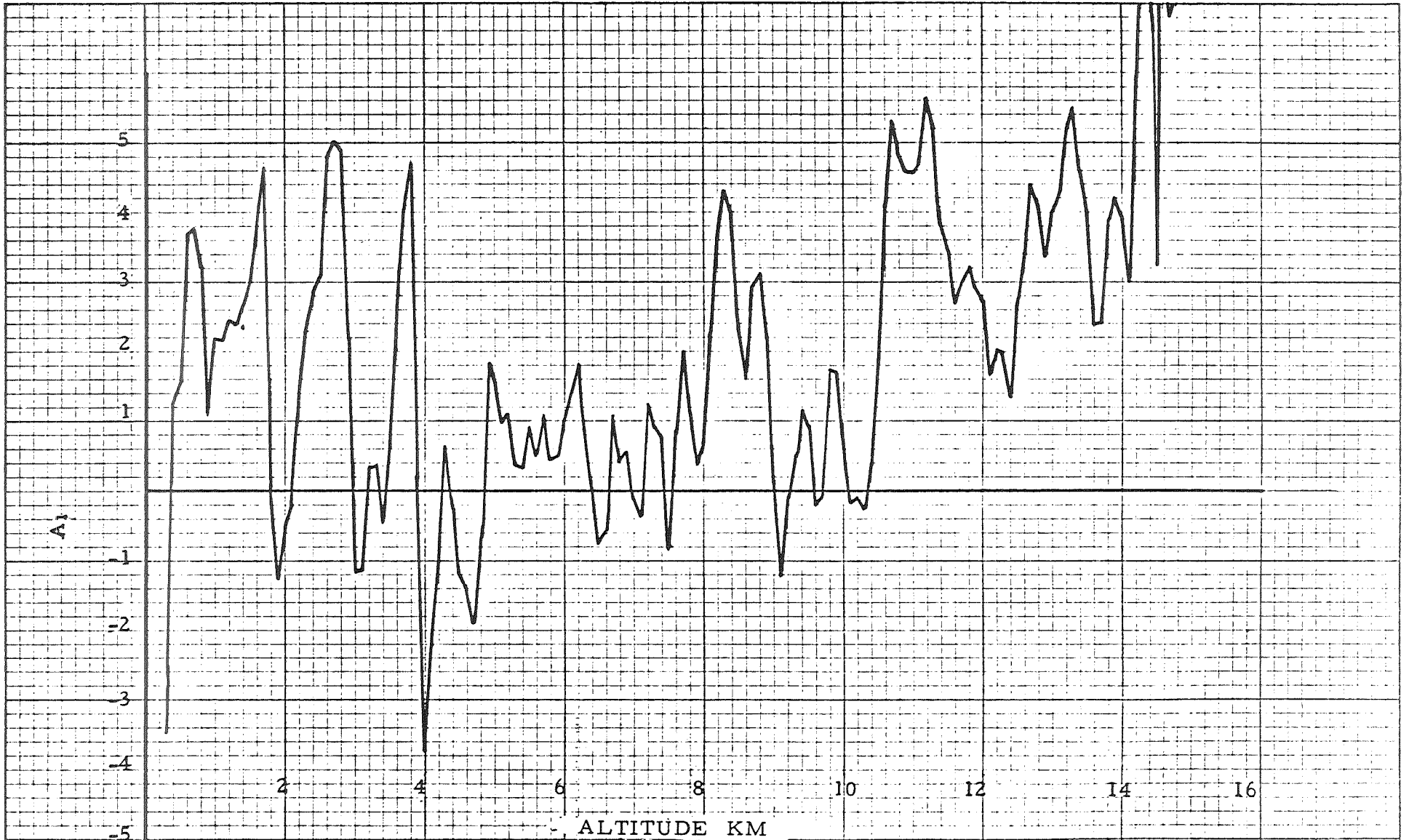
No effort has been made to utilize the coefficients in an analog computer program, as was done during the development program CR846. No data smoothing was performed on the covariance data to insure smooth curves of  $A\phi$ ,  $A1$ ,  $B\phi$  and  $B1$ , rather, an effort was made to retain as much of the natural shape of the covariance as possible.

The covariance matrix will not be presented in this report due to the number of pages that would be required to fully represent all values utilized.



$A_0$  COEFFICIENTS

Figure 6



$A_1$  COEFFICIENTS

Figure 7



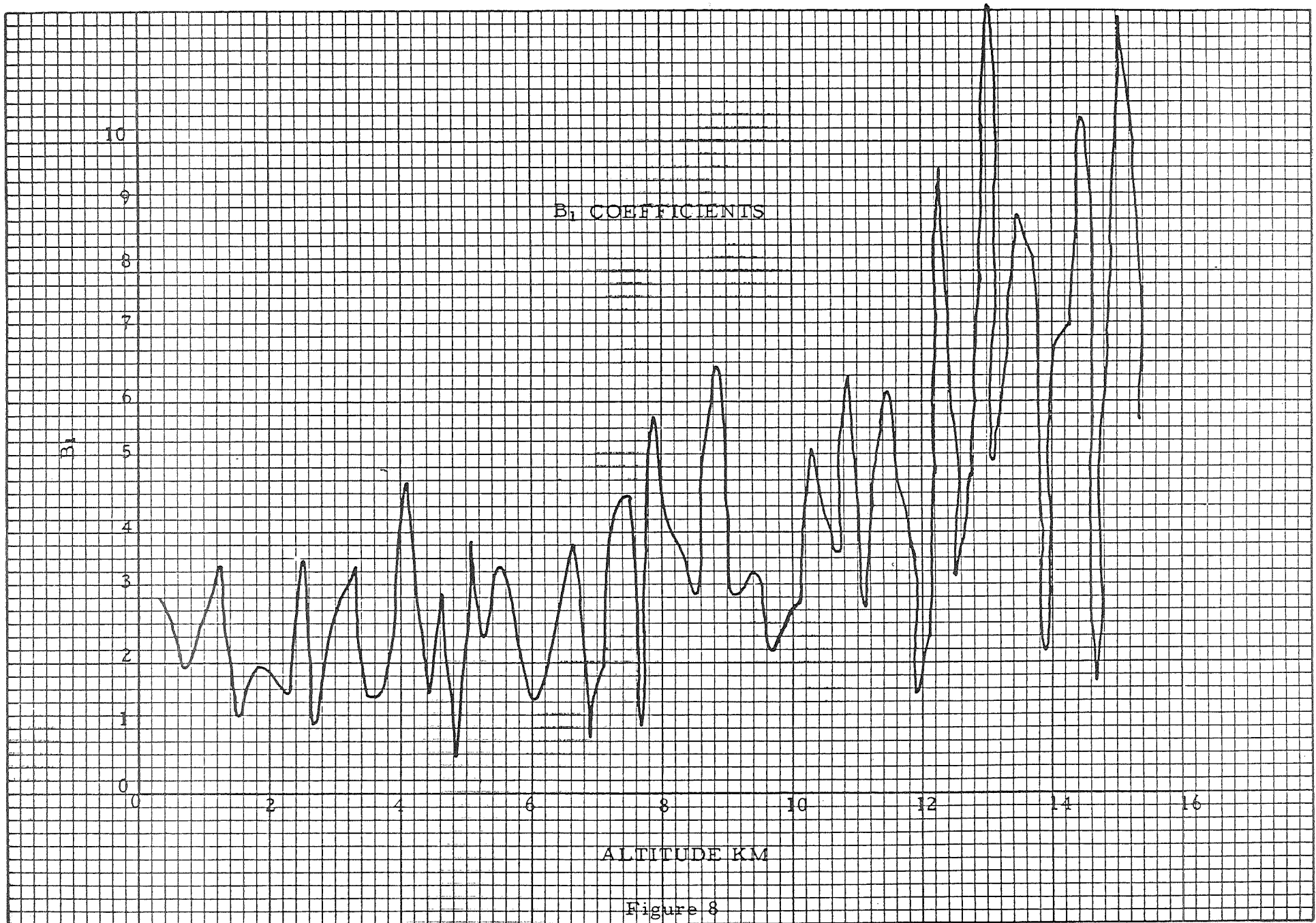


Figure 8



Figure 9

Wind Model	$A_0(h)$	$A_1(h)$	$B_0(h)$	$B_1(h)$
coefficients	0.943709E	00-0.211279E	01 0.317744E	03 0.325791E
beginning	0.802144E	00 0.157822E	01 0.102805E	03 0.233626E
400 meters	0.828959E	00 0.346543E	01 0.279399E	01 0.622888E
and increas-	0.843599E	00 0.369723E	01 0.517014E	01 0.280986E
ing by 50	0.869565E	00 0.380628E	01 -0.196222E	01 0.261020E
meter steps.	0.888715E	00 0.378451E	01 -0.133538E	02 0.320822E
	0.893160E	00 0.352549E	01 -0.649815E	02 0.969438E
	0.879876E	00 0.320951E	01 -0.209659E	03 0.318848E
	0.108513E	00-0.121552E	01 0.697845E	02 0.191760E
	0.583673E	00 0.100762E	01 0.445114E	02 0.324926E
	0.744088E	00 0.236039E	01 -0.228122E	01 0.147955E
	0.682485E	00 0.218749E	01 -0.257396E	01 0.278551E
	0.631291E	00 0.208067E	01 0.320753E	01 0.226791E
	0.614490E	00 0.218301E	01 0.376675E	01 0.180602E
	0.606613E	00 0.234383E	01 0.511350E	01 0.392960E
	0.589931E	00 0.245439E	01 -0.579306E	00 0.116943E
	0.538930E	00 0.239308E	01 0.334284E	00 0.296740E
	0.494154E	00 0.239092E	01 0.754026E	00 0.214086E
	0.452656E	00 0.240968E	01 0.714017E	01 0.217995E
	0.454203E	00 0.265291E	01 0.696000E	01 0.229465E
	0.447347E	00 0.286617E	01 0.954346E	01 0.358303E
	0.430088E	00 0.303655E	01 0.383235E	01 0.961039E
	0.419509E	00 0.325960E	01 0.290048E	02 0.308477E
	0.429190E	00 0.367066E	01 0.467322E	02 0.254762E
	0.505635E	00 0.458601E	01 0.552612E	00 0.117103E
	0.448293E	00 0.464117E	01 0.166026E	03 0.255130E
	0.147697E	00 0.283752E	01 0.707506E	02 0.175903E
	-0.231535E	00 0.398151E	-01 0.149830E	02 0.202026E
	-0.429863E	00-0.174198E	01 -0.117261E	01 0.174727E
	-0.375097E	00-0.126482E	01 -0.863995E	01 0.398003E
	-0.315105E	00-0.639294E	00 0.163908E	01 0.138501E
	-0.300412E	00-0.516356E	00 -0.518522E	01 0.245332E
	-0.263514E	00-0.571142E	-01 -0.887778E	01 0.188162E
	-0.251260E	00-0.200119E	00 0.177309E	02 0.227703E
	-0.174032E	00 0.105719E	01 0.336270E	01 0.147708E
	-0.147314E	00 0.150594E	01 0.974294E	00 0.202946E
	-0.136513E	00 0.164821E	01 0.124524E	02 0.180706E
	-0.118806E	00 0.229773E	01 0.984760E	01 0.245956E
	-0.107625E	00 0.270200E	01 0.224566E	01 0.186950E
	-0.948648E	-01 0.285227E	01 0.301852E	01 0.114634E
	-0.815233E	-01 0.313020E	01 -0.263678E	01 0.149140E
	-0.572561E	-01 0.308738E	01 0.769665E	02 0.262379E
	-0.106617E	00 0.460042E	01 0.580705E	01 0.200069E
	-0.890381E	-01 0.478895E	01 0.746592E	01 0.236154E
	-0.671857E	-01 0.499752E	01 -0.226526E	01 0.356677E
	-0.378602E	-01 0.501886E	01 -0.121078E	02 0.227820E
	-0.298590E	-02 0.480692E	01 0.700622E	00 0.100238E
	0.161719E	-01 0.490334E	01 0.131163E	01 0.335748E
	0.488736E	-01 0.436243E	01 0.423120E	02 0.100214E
	0.101570E	00 0.215354E	01 0.309789E	02 0.180776E
	0.116154E	00 0.601752E	-01 0.841065E	01 0.298941E
	0.986002E	-01-0.117030E	01 -0.409474E	01 0.187342E
	0.686666E	-01-0.923914E	00 0.182336E	01 0.215730E
	0.389109E	-01-0.114997E	01 0.925683E	00 0.228207E
	0.134006E	-01-0.471046E	00 0.793175E	01 0.279629E
	-0.689258E	-02 0.338095E	00 0.735691E	00 0.142143E
	-0.314399E	-01 0.771772E	00 -0.220435E	01 0.304047E
	-0.545792E	-01 0.368309E	00 0.213950E	01 0.246462E
	-0.676400E	-01-0.309014E	00 0.112275E	03 0.438917E
	-0.800424E	-01-0.428428E	00 0.126323E	01 0.209116E
	-0.938338E	-01 0.380518E	-01 0.230223E	02 0.347595E
	-0.112281E	00 0.655090E	00 0.384108E	02 0.214221E
	-0.157654E	00 0.199363E	01 0.101884E	02 0.816395E

-0.199861E 00 0.281284E 01 0.393311E 02 0.345138E 01  
 -0.243638E 00 0.346806E 01 0.184097E 02 0.153076E 01  
 -0.293965E 00 0.405991E 01 0.144685E 02 0.123853E 01  
 -0.354257E 00 0.470163E 01 0.164699E 01 0.240846E 01  
 -0.381615E 00 0.473598E 01 -0.222885E 03 0.232406E 01  
 -0.292293E 00 0.301283E 01 0.937383E 02 0.153164E 01  
 -0.104789E 00 0.159907E 00 0.270512E 02 0.513614E 00  
 0.956720E-01-0.249995E 01 0.125484E 02 0.173146E 01  
 0.203343E 00-0.373976E 01 -0.244337E 02 0.298146E 01  
 0.130900E 00-0.263270E 01 -0.369059E 01 0.193461E 01  
 0.109123E 00-0.223724E 01 -0.576640E 00 0.118695E 01  
 0.707732E-01-0.169739E 01 -0.102937E 01 0.193471E 01  
 0.165342E-01-0.101622E 01 0.537357E 01 0.145935E 01  
 -0.914965E-01 0.267229E 00 0.323680E 02 0.472098E 01  
 -0.119734E 00 0.638621E 00 -0.961770E 00 0.274235E 01  
 -0.669824E-01 0.124399E 00 0.166821E 00 0.207993E 01  
 -0.268649E-01-0.252290E 00 0.129107E 01 0.249531E 01  
 -0.928986E-02-0.428739E 00 0.108764E 02 0.277480E 01  
 0.652044E-01-0.116330E 01 0.130569E 01 0.210529E 01  
 0.851464E-01-0.135310E 01 -0.551410E-01 0.805345E 00  
 0.955754E-01-0.141925E 01 0.431789E 00 0.676609E 00  
 0.112430E 00-0.155327E 01 0.124526E 01 0.154210E 01  
 0.153925E 00-0.188800E 01 0.364275E 00 0.156582E 01  
 0.985109E-01-0.120286E 01 0.109198E 01 0.104092E 01  
 0.534475E-01-0.666931E 00 0.206858E 02 0.190134E 01  
 -0.554763E-01 0.436444E 00 0.258248E 02 0.314967E 01  
 -0.176818E 00 0.156574E 01 0.308060E 00 0.356925E 00  
 -0.195050E 00 0.180132E 01 -0.251288E 01 0.256123E 01  
 -0.158473E 00 0.154293E 01 -0.328275E 01 0.329153E 01  
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 -0.727124E-01 0.944011E 00 -0.159492E 00 0.390000E 00  
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 0.926723E-02 0.351162E 00 -0.282795E 00 0.137870E 01  
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 -0.701579E-02 0.327856E 00 0.269266E 01 0.267354E 01  
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 -0.883603E-01 0.918793E 00 0.228752E 00 0.565152E 00  
 -0.874392E-01 0.812315E 00 -0.319728E 01 0.190258E 01  
 -0.634038E-01 0.466120E 00 0.182124E 00 0.178538E 01  
 -0.893636E-01 0.623129E 00 0.289520E 00 0.339844E 00  
 -0.142388E 00 0.106653E 01 0.456813E 01 0.451491E 01  
 -0.129098E 00 0.848542E 00 0.955779E 01 0.300561E 01  
 -0.877656E-01 0.412275E 00 -0.649456E 01 0.312864E 01  
 -0.568658E-01 0.137209E 00 0.162416E 01 0.321392E 01  
 -0.877489E-01 0.485996E 00 0.146184E 01 0.266874E 01  
 -0.111217E 00 0.752009E 00 0.807066E 00 0.936857E 00  
 -0.133013E 00 0.100761E 01 0.115610E 01 0.990456E 00  
 -0.161121E 00 0.129956E 01 0.613590E 00 0.176255E 01  
 -0.171295E 00 0.142216E 01 0.471102E 01 0.270732E 01  
 -0.203459E 00 0.173446E 01 0.130343E 00 0.706038E 00  
 -0.207558E 00 0.180690E 01 -0.426128E 00 0.252709E 01  
 -0.152783E 00 0.138752E 01 0.193483E 01 0.137393E 01  
 -0.881863E-01 0.863871E 00 -0.411499E 02 0.214294E 01  
 -0.287331E-01 0.375039E 00 -0.991124E 00 0.398373E 01  
 0.162784E-01-0.550087E-02 0.680920E 00 0.852164E 00  
 0.668823E-01-0.403240E 00 0.326638E 01 0.278390E 01  
 0.110078E 00-0.751574E 00 0.193382E 01 0.313986E 01  
 0.151361E 00-0.109838E 01 -0.277548E 01 0.123052E 01  
 0.999856E-01-0.576802E 00 0.152623E 02 0.314059E 01  
 0.357825E-01 0.119017E 00 0.624935E 01 0.324824E 01  
 -0.473252E-01 0.107978E 01 0.995210E 00 0.343629E 01  
 -0.164699E-01 0.804549E 00 0.730338E-01 0.248308E 01  
 0.230375E-01 0.433262E 00 -0.185026E 01 0.106137E 01  
 0.445789E-01 0.244383E 00 0.101309E 01 0.378020E 01

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 0.814008E-01 -0.369683E 00 0.848892E 01 0.191606E 01  
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 0.401461E-02 0.111611E 00 0.201698E 02 0.422955E 01  
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 -0.145870E 00 0.322869E 01 -0.266944E 00 0.372284E 00  
 -0.736722E-01 0.227735E 01 -0.932379E 01 0.110032E 01  
 -0.715723E-02 0.127533E 01 0.330320E 01 0.115291E 01  
 -0.315233E-01 0.160231E 01 0.197694E 02 0.278054E 01  
 -0.617113E-01 0.211575E 01 0.310139E 02 0.280087E 01  
 -0.101106E 00 0.293162E 01 0.540204E 01 0.156713E 01  
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 -0.569459E-01 0.474103E 00 0.213976E 01 0.318600E 01  
 -0.727367E-01 0.724660E 00 0.278712E 01 0.273670E 01  
 -0.949331E-01 0.114677E 01 -0.438580E 01 0.510471E 01  
 -0.857830E-01 0.985494E 00 0.202020E 01 0.287852E 01  
 -0.827649E-01 0.913476E 00 -0.344831E 01 0.226104E 01  
 -0.516514E-01 0.457636E 00 0.273400E 00 0.435788E 00  
 -0.848176E-03 -0.203705E 00 0.620077E 01 0.256575E 01  
 0.196514E-01 -0.402445E 00 0.741951E 00 0.346681E 01  
 0.280312E-02 -0.108226E 00 0.281501E 02 0.364394E 01  
 -0.350578E-01 0.687286E 00 0.993534E 01 0.446738E 01  
 -0.887074E-01 0.171410E 01 -0.189468E 02 0.323725E 01  
 -0.440592E-01 0.113393E 01 0.273308E 01 0.224143E 01  
 -0.726798E-01 0.166357E 01 -0.130668E 01 0.203472E 01  
 -0.347394E-01 0.112260E 01 -0.140451E 01 0.257482E 01  
 -0.803917E-02 0.726241E 00 -0.109383E 02 0.287472E 01  
 0.390208E-02 0.508030E 00 0.874653E 01 0.259108E 01  
 0.416689E-01 -0.167922E 00 0.874102E 01 0.892755E 00  
 0.843589E-01 -0.965042E 00 0.301740E 01 0.265774E 01

0.416532E-01	-0.103511E	00	-0.147324E	02	0.339560E	01
0.842536E-02	0.339265E	00	0.214469E	01	0.289133E	01
0.345767E-01	-0.254715E	00	-0.351384E	01	0.285443E	01
0.258097E-01	-0.127857E	00	0.563259E	01	0.343492E	01
-0.313023E-02	0.390412E	00	0.110062E	02	0.255603E	01
-0.292308E-01	0.820610E	00	0.284168E	02	0.522755E	01
-0.876366E-01	0.189888E	01	-0.110463E	02	0.164073E	01
-0.788873E-01	0.163347E	01	0.363614E	02	0.663775E	00
-0.219552E	0.410369E	01	0.924857E	02	0.530615E	01
-0.267449E	0.490261E	01	0.340276E	02	0.422478E	01
-0.287308E	0.531620E	01	-0.702180E	01	0.832201E	01
-0.292084E	0.527293E	01	-0.338406E	02	0.443831E	01
-0.277724E	0.480535E	01	-0.170577E	02	0.279550E	01
-0.270394E	0.448562E	01	0.334339E	01	0.365841E	01
-0.271627E	0.456470E	01	-0.137689E	02	0.463350E	01
-0.263149E	0.442933E	01	0.409989E	01	0.311957E	01
-0.265333E	0.457145E	01	-0.131566E	03	0.579451E	01
-0.205690E	0.362649E	01	0.147910E	03	0.641682E	01
-0.235240E	0.473098E	01	0.287904E	02	0.325106E	01
-0.227334E	0.514200E	01	0.366364E	02	0.376399E	01
-0.227942E	0.563281E	01	-0.281987E	01	0.167123E	01
-0.185201E	0.561348E	01	-0.212245E	02	0.261058E	01
-0.112666E	0.519499E	01	-0.247223E	02	0.286463E	01
-0.324127E-01	0.460507E	01	-0.806165E	01	0.825717E	00
0.491175E-01	0.383808E	01	-0.311395E	02	0.305179E	01
0.115468E	0.325733E	01	0.526248E	01	0.472477E	01
0.153479E	0.346851E	01	-0.556439E	02	0.646884E	01
0.202516E	0.277998E	01	-0.709221E	01	0.377810E	01
0.221717E	0.268193E	01	0.142144E	01	0.356651E	01
0.235652E	0.275793E	01	0.181530E	02	0.610192E	01
0.261008E	0.302703E	01	0.118298E	01	0.314013E	01
0.278677E	0.309844E	01	0.358224E	01	0.320618E	01
0.297059E	0.323103E	01	-0.289945E	01	0.182932E	01
0.307888E	0.314721E	01	-0.120120E	02	0.440774E	01
0.306443E	0.292469E	01	-0.127094E	02	0.278675E	01
0.288999E	0.259618E	01	0.112091E	01	0.309122E	01
0.303658E	0.273263E	01	-0.127026E	03	0.102252E	02
0.186435E	0.133410E	01	0.277612E	02	0.150097E	02
0.231985E	0.164275E	01	0.376237E	01	0.785726E	01
0.247574E	0.175129E	01	0.758515E	01	0.486343E	01
0.272628E	0.203112E	01	-0.109227E	02	0.445193E	01
0.251562E	0.182161E	01	0.256207E	01	0.362642E	01
0.264645E	0.198973E	01	-0.391174E	01	0.272127E	01
0.231318E	0.173206E	01	-0.107253E	02	0.327157E	01
0.187696E	0.134694E	01	0.504668E	01	0.264535E	01
0.191121E	0.151334E	01	-0.234074E	02	0.951927E	01
0.266658E	0.269114E	01	-0.234022E	03	0.417892E	01
0.785594E-01	0.555499E	00	0.489014E	03	0.903209E	01
0.280812E	0.337952E	01	0.635261E	02	0.515073E	01
0.334078E	0.403810E	01	0.185508E	02	0.331071E	01
0.372052E	0.438840E	01	-0.594028E	01	0.271704E	01
0.366775E	0.428111E	01	-0.888763E	01	0.255884E	01
0.353632E	0.410568E	01	-0.615656E	01	0.651342E	01
0.364029E	0.405692E	01	-0.532421E	02	0.588143E	01
0.295977E	0.336630E	01	-0.269034E	02	0.478814E	01
0.290954E	0.316892E	01	0.101105E	02	0.283401E	01
0.442194E	0.400850E	01	-0.400098E	03	0.958163E	01
0.288827E	0.248105E	01	0.513667E	03	0.126596E	02
0.580688E	0.429913E	01	0.106781E	03	0.816671E	01
0.695055E	0.488798E	01	0.563036E	02	0.745197E	01
0.774086E	0.522821E	01	0.340374E	02	0.685092E	01
0.847600E	0.547511E	01	-0.476597E	01	0.513980E	01
0.906111E	0.548923E	01	-0.111494E	02	0.166127E	01
0.941966E	0.520077E	01	-0.107618E	03	0.870726E	01
0.939967E	0.455639E	01	-0.591021E	02	0.671453E	01
0.947013E	0.410481E	01	-0.146933E	02	0.684433E	01

0.968542E 00 0.397681E 01 -0.170658E 03 0.118796E 02  
 0.831059E 00 0.266810E 01 -0.332233E 02 0.951173E 01  
 0.786933E 00 0.237258E 01 0.839044E 01 0.848803E 01  
 0.798301E 00 0.245338E 01 -0.282529E 01 0.883469E 01  
 0.796542E 00 0.241754E 01 0.207576E 03 0.101573E 02  
 0.101496E 01 0.342709E 01 0.627974E 02 0.644307E 01  
 0.111458E 01 0.387480E 01 0.524840E 02 0.982513E 01  
 0.112336E 01 0.406166E 01 0.404784E 02 0.820096E 01  
 0.110866E 01 0.422185E 01 0.121770E 02 0.847326E 01  
 0.988621E 00 0.416457E 01 -0.176658E 04 0.433247E 01  
 0.770002E 00 0.389730E 01 -0.406168E 04 0.172576E 02  
 0.657279E 01 0.179491E 02 -0.634116E 04 0.223200E 02  
 0.433429E -02 0.304213E 01 0.787306E 03 0.162744E 02  
 0.904204E 00 0.492957E 01 0.439958E 03 0.878639E 01  
 0.867487E 00 0.597022E 01 0.158979E 04 0.783944E 01  
 0.200934E 01 0.106794E 02 -0.635603E 03 0.696089E 01  
 0.162938E 01 0.103614E 02 0.287901E 05 0.104156E 02  
 0.366376E 01 0.209056E 02 0.381437E 05 0.613517E 01  
 0.757446E 01 0.461476E 02 0.407709E 05 0.857945E 01  
 0.819826E 01 0.518227E 02 -0.760323E 05 0.711271E 01  
 0.553436E 00 0.221404E 01 0.493831E 05 0.142592E 02  
 0.381869E 01 0.275444E 02 0.522609E 03 0.935415E 01  
 0.421557E 01 0.298740E 02 0.633534E 03 0.389944E 01  
 0.299537E 01 0.178853E 02 -0.166736E 04 0.103369E 02  
 0.920998E 00 0.685388E 01 -0.178671E 04 0.616817E 01  
 0.106475E 01 0.487155E 01 0.831537E 04 0.737039E 01  
 0.250372E 01 0.165377E 02 0.324611E 04 0.710284E 01  
 0.337774E 01 0.196197E 02 -0.102391E 05 0.169322E 01  
 0.482319E 01 0.292515E 02 -0.145028E 06 0.480233E 01  
 -0.144521E 01 -0.676340E 01 -0.413870E 05 0.428825E 01  
 0.166828E 02 0.762554E 02 -0.110797E 06 0.103385E 02  
 0.441480E 01 0.143055E 02 0.102068E 05 0.166851E 02  
 0.611192E 01 0.282888E 02 -0.247734E 05 0.957319E 01  
 0.569515E 00 0.109599E 01 0.170902E 05 0.886191E 01  
 0.381050E 01 0.112204E 02 -0.744473E 04 0.960273E 01  
 0.878811E 01 0.330381E 02 -0.263733E 05 0.101284E 02  
 0.131514E 01 0.540148E 01 0.111555E 05 0.915977E 01  
 0.438483E 01 0.135823E 02 0.111862E 05 0.835442E 01  
 0.667309E 01 0.258263E 02 -0.850298E 05 0.104735E 02  
 -0.233651E 00 0.200355E 00 -0.740939E 04 0.563203E 01  
 0.990260E 01 0.321878E 02 -0.148469E 05 0.100922E 02  
 0.162218E 01 0.543645E 01 0.237776E 04 0.114209E 02  
 0.205831E 01 0.989410E 01 0.265788E 04 0.902429E 01  
 -0.908996E 00 0.120616E 00 0.717033E 02 0.855989E 01  
 -0.217763E 01 -0.332186E 01 -0.113591E 02 0.757271E 01

## VIII. COMPUTER PROGRAMS

The computer programs furnished are the programs to calculate the covariance matrix,  $A\phi$ ,  $A1$ ,  $B\phi$  and  $B1$ . The computer program to calculate the derivatives by polynomial curve fit are not furnished since this routine is standard in computation libraries and depends on the researchers preferences. The first program consisting of three pages is the computer program for calculating the covariance matrix. The second program consisting of one page is the program to calculate the A coefficients. The third program consists of one page and is the program to develop the B coefficients. The covariance program is developed for the IBM 7094 computer whereas the A and B coefficient programs were written for the IBM 360 computer.



## COMPUTER SYMBOLS

AV1	Average wind speed
AV2	
AVW	
AVS	
AVPIJ	Product of the average wind speed at altitudes I, J
CORR	Correlation of wind speeds at altitudes I, J
COV	Covariance of wind speeds at altitudes I, J
C	Covariance
DC	Derivative of covariance
DDC	Second Derivative of covariance
IT	Tape Numbers
S	Sum
W(I, K)	Wind velocity at altitude I, record K

## MAIN2 - EFN SOURCE STATEMENT - IFN(S) -

```

      DIMENSION W(3,125), AVW(2), COV(2,350), CORR(3),
1    CORR5(2)
1    FORMAT(IH)
2    FORMAT(F10.1 / (10F13.4))
4    FORMAT(8F10.4)
      DO 100 IB=36,159
          IB2=2*IB
          IF (IB=40) 202,202,200
200  IF (IB=80) 203,203,201
201  IF (IB=120) 204,204,205
202  IT=8
          ISKIP=2*(IB-1)
          GO TO 206
203  IT=9
          ISKIP=2*(IB-4)
          GO TO 206
204  IT=10
          ISKIP=2*(IB-8)
          GO TO 206
205  IT=11
          ISKIP=2*(IB-12)
206  IF (ISKIP) 209,209,207
207  DO 208 I=1,ISKIP
208  READ(IT)
209  READ(IT) (W(I,I),I=1,125)
          READ(IT) (W(2,I),I=1,125)
          WRITE(6,1)
          I1=0
          I2=0
          AV1=0.
          AV2=0.
          DO 14 K=2,125
              IF (W(1,K)) 12,11,12
11  I1=I1+1
              GO TO 13
12  AV1=AV1+W(1,K)
13  IF (W(2,K)) 132,131,132
131  I2=I2+1
              GO TO 14
132  AV2=AV2+W(2,K)
14  CONTINUE
          I1=1250-I1
          I2=1250-I2
          AI=I1
          AVW(1)=AV1/AI
          AI=I2
          AVW(2)=AV2/AI
          AVP11=AVW(1)*AVW(1)
          AVP12=AVW(1)*AVW(2)
          AVP22=AVW(2)*AVW(2)
          DO 22 I=1,3
              CORR(I)=0.
              K=0
              IF (I=2) 15,16,17
15  L=1

```

MAIN2      -    EFN    SOURCE STATEMENT    -    IFN(S)    -

12/12

```

M=1
GO TO 18
16 L=1
M=2
GO TO 18
17 L=2
M=2
18 DO 21 J=2,1251
COR=W(L,J)*W(M,J)
IF(COR) 20,19,20
19 K=K+1
GO TO 21
20 CORR(I)=COR+CORR(I)
21 CONTINUE
K=1250-K
AI=K
CORR(I)=CORR(I)/AI
22 CONTINUE
COV11=CORR(1)=AVP11
COV12=CORR(2)=AVP12
COV22=CORR(3)=AVP22
ALT=W(1,1)
ALT2=W(2,1)
ISF=320=IB2
DO 90 IS=1,ISF
ITEST=IB2+IS
IF(ITEST=80) 216,216,2100
2100 CALL CLOSE (8,4)
CALL REW08
210 IF(ITEST=160) 213,213,2110
2110 CALL CLOSE (9,4)
CALL REW09
211 IF(ITEST=240) 214,214,2150
2150 CALL CLOSE (10,4)
CALL REW10
GO TO 215
213 IT=9
GO TO 216
214 IT=10
GO TO 216
215 IT=11
216 READ(IT) (W(3,I),I=1,1251)
J=0
AVS=0.
DO 25 I=2,1251
IF(W(3,I)) 24,23,24
23 J=J+1
GO TO 25
24 AVS=AVS+W(3,I)
25 CONTINUE
J=1250-J
AI=J
AVS=AVS/AI
DO 30 I=1,2
CORR(I)=0.
K=0

```

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MAIN2 - EFN SOURCE STATEMENT - IFN(S) -

```
DO 29 J=2,1251
CORS=W(1,J)*W(3,J)
IF(CORS) 27,26,27
26 K=K+1
GO TO 29
27 CORRS(1)=CORS*(1)+CORS
29 CONTINUE
K=1250-K
AI=K
CORRS(1)=CORRS(1)/AI
30 CONTINUE
AVP31=AVW(1)*AVS
AVP32=AVW(2)*AVS
COV(1,IS)=CORRS(1)-AVP31
COV(2,IS)=CORRS(2)-AVP32
90 CONTINUE
WRITE(6,2) ALT,AVW(1),COV11,COV12,(COV(1,1),I=1,ISF)
WRITE(7,4) ALT,AVW(1),COV11,COV12,(COV(1,1),I=1,ISF)
WRITE(6,1)
WRITE(6,2) ALT2,AVW(2),COV22,(COV(2,1),I=1,ISF)
WRITE(7,4) ALT2,AVW(2),COV22,(COV(2,1),I=1,ISF)
CALL CLOSE(11,4)
CALL REWII
100 CONTINUE
STOP
END
```

```

C      A COEFFICIENT PROGRAM
      DIMENSION AU%320<,A1%320<,C%320<,DC%320<,DDC%320<
C      S=SUM C=COV,DC=DERIVATIVES OF COV
3      FORMAT(1H1)
4      FORMAT(6E15.7)
5      FORMAT(5E15.7)
6      FORMAT(1X9H A0 COEFF  )
7      FORMAT(1X9H A1 COEFF  )
8      FORMAT%40X110 <
      READ(7)
      I1#10
      DO 40 I=2,I1
      WRITE%3,8< I
      READ( 7 )(C(L),DC(L),DDC(L),L=1,I)
      SCC=0.
      SCDC=0.
      SCDDC=0.
      SDCDC=0.
      SDCDDC=0.
      DO 15 J=1,I
      P=C(J)
      SCC=C(J)*P+SCC
      SCDC=DC(J)*P+SCDC
15     SCDDC=DDC(J)*P+SCDDC
      DO 30 J=1,I
      P=DC(J)
      SDCDC=DC(J)*P+SDCDC
30     SDCDDC=DDC(J)*P+SDCDDC
      TOP1=-SCC*SDCDDC+SCDC*SCDDC
      BOT1=SCC*SDCDC-SCDC*SCDC
      TOP2=-SCDDC*SDCDC+SCDC*SDCDDC
      BOT2=BOT1
      A1(1)=TOP1/BOT1
      WRITE%3,3<
      WRITE%3,9< SCC,SCDC,SCDDC,SDCDC,SDCDDC,TOP1,BOT1,TOP2,BOT2
9      FORMAT%///10E13.5<
      WRITE%3,10< %C%L<,DC%L<,DDC%L<,L#1,I<
10     FORMAT%//%3E20.7<<
40     AU(1)=TOP2/BOT2
      WRITE(3,3)
      WRITE(3,6)
      WRITE(3,4) (AU(J),J=1,I1)
      WRITE(2,5) (AU(J),J=1,I1)
      WRITE(3,3)
      WRITE(3,7)
      WRITE(3,4) (A1(J),J=1,I1)
      WRITE(2,5) (A1(J),J=1,I1)
      STOP
      END

```

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/JOB SINGLE,GO  
BEGIN COMPILATION

```

C      PROGRAM TO CALCULATE BU,B1
S.0001  DIMENSION AU%320<,A1%320<,BU%320<,B1%320<,DSTARC%320<,
        IDC%320<,G%3<,C%320<
S.0002  IALL#306
S.0003  READ%1,1<%AU%JK,J#1,IALL<
S.0004  READ%1,1<%A1%JK,J#1,IALL<
S.0005  WRITE(3,2) (AU(J),A1(J),J=1,IALL)
S.0006  1 FORMAT(60X E15.7/(5E15.7))
S.0007  READ%1,7<%DSTARC%JK,J#1,IALL<
S.0008  7 FURMAT%6E12.6<
S.0009  READ%1,2<%DC%JK,J#1,IALL<
S.0010  WRITE(3,2) (DSTARC(J),DC(J),J=1,IALL)
S.0011  2 FURMAT%6E13.6<
S.0012  DO 10 I#1,IALL
S.0013  J#I&1
S.0014  K#J&1
S.0015  CON1#AU%JK-2.*A1%K<
S.0016  CON2#A1%JK
S.0017  CON3#AU%I<-2.*A1%I<&A1%JK
S.0018  CON4#AU%JK&2.*A1%K<
S.0019  CON5#A1%JK-AO%K<-2.*A1%K<
S.0020  ALFA=ABS(DSTARC(1)-DC(1))
S.0021  WRITE(3,2) CON1,CON2,CON3,CON4,CON5,ALFA
S.0022  5 G%I<=SQRT%ALFA<
S.0023  G%K<#%CON3*CON4-CON1*CON2/CON2*CON4-CON1*CON5<*G%I<
S.0024  G%J<#%CON2*G%I<-CON5*G%K<</CON4
S.0025  B1(1)=G(I)
S.0026  BU%I<#.5*%-3.*G%I<&4.*G%J<-G%K<<&A1%I<#*B1%I<
S.0027  10 CONTINUE
S.0028  WRITE%3,8<
S.0029  8 FORMAT%1H1<
S.0030  WRITE%3,9<
S.0031  9 FORMAT%7X2HB012X2HB1<
S.0032  WRITE%3,6<%BU%I<,b1%I<,1#1,IALL<
S.0033  6 FORMAT(20X2E13.6)
S.0034  STOP
S.0035  END

```

## CRITIQUE OF RESULTS

The results obtained, as shown in Figures 6, 7, 8, and 9, show an unusual amount of variability. This variability appears to be directly proportional to the variability in the covariance derivatives,  $\frac{dC}{dh}$ ,  $\frac{d^2 C}{dh^2}$ . An important conclusion results from this proportionality: that this technique will not produce accurate results on turbulence data. A cursory view of turbulence data obtained by separating turbulence frequencies from Jimsphere records reveals the impracticality in attempting to obtain these derivatives. To obtain accurate results on the filter development, accurate derivatives are required. This accuracy is difficult to obtain using polynomial techniques; however, any other technique results in smoothing of the data, thereby losing the effect of the selection of an appropriate lag size. Lag size was discussed in Section III.

The dependence on covariance derivative variability can be seen in Figures 6, 7, 8, and 9. The data up to 14000 meters were segmented for curve fitting every 2000 meters (500 meters between points), introducing some smoothing in the data. The data above 14000 meters were curve fit at 250 meter increments (50 meters between each point). The latter case detects each change of sign in the derivatives along the span of the covariance, resulting in the degree of variability as shown above 14000 meters; whereas, the former maintains the same derivative sign for large increments due to some smoothing introduced by the 2000 meter increment selection, resulting in less variability as shown below 14000 meters. A case for highly smoothed data which exhibits no variability can be seen in CR846.

The values of the covariance matrix of the Jimsphere data obtained during this analysis appears to be accurate. The data sample was adequate and sufficient care was taken such that the present matrix has the necessary elements (accuracy, sample size, etc.) such that validity can be placed on its use in further analyses.

Attempts were made during the present analysis to expand the order of the differential equation to an order greater than two (2). These were unsuccessful, primarily due to the order of the right hand side of Equation (1), Section II.

The coefficients of the right hand side were developed from results obtained by Solodovnikov<sup>(4)</sup>, based on the works of Batkov<sup>(3)</sup>, particularly Solodovnikov's Equations (10.27), page 383 and (10.60), page 390, Reference (4). The major difficulty that arises was pointed out by Stear<sup>(7)</sup>, that the Equation (10.27) is not a recursive relationship as stated by Batkov. Further, as became

evident during the analysis, Solodovnikov's Equation (10.60) puts a restraint on the data, forcing the value of  $\frac{dC(h_i, h_j)}{dh_j}$   $\left| \begin{array}{c} - \\ - \\ h_i = h_j \end{array} \right| \frac{dC(h_i, h_j)}{dh_j} \left| \begin{array}{c} - \\ - \\ h_i = h_j \end{array} \right| +$

to be positive. When analysing the data, it is seen that there is no rational basis for such a constraint.

Equation (10.60) is equal to the value of the difference between the derivatives at  $h_i = h_j$ , or  $(-1)^{n-m-1} \frac{b^2_m(t)}{a^2_n(t)}$ , the value of the discontinuity at  $h_i = h_j$ , which has meaning when viewed as the initial value of  $g(h_i, h_j)$  at  $h_i = h_j$ . When considered as the initial value of  $g(h_i, h_j)$ , (all values of  $g(h_i, h_j)$  for  $h_i > h_j = 0$ ), only the magnitude of Equation (10.60) has significance. Therefore, this report utilized the magnitude of Equation (10.60).



## CONCLUSIONS

1. The shaping filter can be developed using a large volume of winds and turbulence data.
2. The Jimsphere data handling requires a computer with large storage capacity.
3. Floating point polynomial fitting of the covariance curves produces derivatives of sufficient accuracy for filter development, except the end points of each segment fit.
4. The technique can be very effective for smaller more tractable data samples, but for large, highly nonstationary processes of high frequency content, accuracy of coefficients becomes critical with small lag sizes.
5. This technique is not readily useable on data of high variability, such as turbulence.

## RECOMMENDATIONS

1. The accuracy of the coefficients should be validated prior to assigning credibility to analyses utilizing the filter coefficients.
2. Consideration should be given to use of this technique with partitions of the covariance matrix.
3. Some smoothing of the covariance matrix should be considered.
4. When using a polynomial curve fit to determine the covariance derivatives, the segments that are curve fit should overlap to eliminate the derivative error on the polynomial end points.

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