

Reacting Fluids Laboratory
Department of Chemical Engineering
Louisiana State University
Baton Rouge, Louisiana

A Technical Report
on
THERMODYNAMIC PROPERTIES IN POLYNOMIAL FORM
FOR CARBON, HYDROGEN, NITROGEN, AND
OXYGEN SYSTEMS FROM 300 TO 15000°K

by

D. D. Esch, Research Associate
A. Siripong, Research Assistant
R. W. Pike, Associate Professor
Co-Principle Investigator

1411-38174/7
(ACCESSION NUMBER)
20
(PAGES)
CR-111989
(NASA CR OR TMX OR AD NUMBER)

FACILITY FORM 602

NASA-RFL-TR-70-3

Prepared under Grant NGR19-001-059

for

Langley Research Center

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

REPRODUCED BY
**NATIONAL TECHNICAL
INFORMATION SERVICE**
U. S. DEPARTMENT OF COMMERCE
SPRINGFIELD, VA. 22161

November 15, 1970

50

N O T I C E

**THIS DOCUMENT HAS BEEN REPRODUCED FROM
THE BEST COPY FURNISHED US BY THE SPONSORING
AGENCY. ALTHOUGH IT IS RECOGNIZED THAT CER-
TAIN PORTIONS ARE ILLEGIBLE, IT IS BEING RE-
LEASED IN THE INTEREST OF MAKING AVAILABLE
AS MUCH INFORMATION AS POSSIBLE.**

Reacting Fluids Laboratory
Department of Chemical Engineering
Louisiana State University
Baton Rouge, Louisiana

A Technical Report
on
THERMODYNAMIC PROPERTIES IN POLYNOMIAL FORM
FOR CARBON, HYDROGEN, NITROGEN, AND
OXYGEN SYSTEMS FROM 300 TO 15000°K

by

D. D. Esch, Research Associate
A. Siripong, Research Assistant
R. W. Pike, Associate Professor
Co-Principal Investigator

NASA-RFL-TR-70-3

Prepared under Grant NGR19-001-059

for

Langley Research Center

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

November 15, 1970

ABSTRACT

Thermodynamic properties of carbon, hydrogen, nitrogen and oxygen compounds as well as selected inerts (Neon, Argon and Helium) are fitted to general polynomial forms over the following temperature ranges: 300 to 1000°K, 1000 to 6000°K, and 6000 to 15000°K. A reference state of 298.16°K and 1.0 atmosphere was chosen with H₂, N₂, O₂, Ne, Ar, He, C (solid) and e⁻ as reference elements. Seven coefficients are tabulated for each of the 99 species considered and for each temperature range of interest.

ACKNOWLEDGMENTS

The authors would like to give acknowledgment to Dean Mayers and James Callender for their helpful assistance in collecting and analyzing the data. Special thanks is also given to Mrs. Sue Mayers for the typing of this report.

NOMENCLATURE

C_p°	heat capacity per mole at constant pressure
F_T°	standard free energy of a pure substance at temperature T
H_T°	standard enthalpy of a pure substance at temperature T
H_0°	standard enthalpy of a pure substance at a temperature of 0°K
R	universal gas constant, 1.987 cal/mole°K
S_T°	standard entropy of a pure substance at temperature T
T	absolute temperature
y_i	mole fraction of species i
$(\Delta H_f^\circ)_T$	standard heat of formation at temperature T

Subscripts:

i	refers to species
T	absolute temperature (°K)
P	constant pressure

Superscripts:

°	denotes the standard state (pure substance at 1 atmosphere pressure)
---	--

TABLE OF CONTENTS

<u>Section</u>	<u>Page</u>
ABSTRACT	ii
ACKNOWLEDGMENTS	iii
NOMENCLATURE	iv
I INTRODUCTION	1
II MATHEMATICAL DEVELOPMENT	2
III DISCUSSION OF RESULTS	8
IV SUMMARY	13
V BIBLIOGRAPHY	15
APPENDIX A. Tabulated Constants for Polynomial Approximation of Thermodynamic Properties	17
APPENDIX B. Determination of $(\Delta H_f^\circ)_{298.16}$ from $(\Delta H_f^\circ)_0$	29
APPENDIX C. Listing of Computer Data	33
APPENDIX D. Tabulated Thermodynamic Properties	40
VI DISTRIBUTION	

I. INTRODUCTION

In recent years it has become increasingly important to accurately determine the equilibrium compositions of multicomponent systems at very high temperatures. In the flow-field adjacent to ablative thermal protection systems, temperatures as high as 30,000°K are not uncommon. The most widely used method for calculating equilibrium compositions for such problems is that of free energy minimization.

In order to make use of this method, the free energy of each of the components is required as a function of temperature. It is convenient, for the computer implementation of free energy minimization techniques, to have this information as polynomial curve-fits. For temperatures below 6000°K, McBride et.al. (Ref. 9) contains such information for many species of interest. These thermodynamic properties are based on a reference state at 298.16°K. There are also numerous reports (Refs. 8 and 12) on high temperature flow field studies in which the free energy of the species of interest for the particular study have been fitted to polynomials, frequently at a reference state of 0°K.

Two alternatives are available for the procurement of data necessary for equilibrium calculations: (1) the investigator is required to search through various reports for the necessary polynomial constants, then transform these values to a consistent reference state or (2) compute the polynomial fits the free energy functions as found in a number of reports, (Refs. 2-6, 11 and 13). Both procedures involve an additional evaluation of the heats of formation for the species of interest.

It is the purpose of this report to consolidate into one source, a standard set of polynomial curve-fits of thermodynamic data for species of particular interest in high temperature studies of ablative thermal

protection systems. In the section to follow, there will be a review of the relationships between the thermodynamic properties and a discussion of the techniques used to obtain the polynomial constants as reported in Appendix A.

II. MATHEMATICAL DEVELOPMENT

In this section we will first derive from fundamental relations a convenient polynomial form for the expression of standard free energy and demonstrate several important relations which permit the evaluation of other thermodynamic properties from the same set of constants. The remainder of the section will contain a discussion of the procedure used in this work to evaluate the polynomial constants.

Evaluation of Polynomial Forms

Thermodynamic data is widely available (Refs. 2-6, 11, 13) for many substances relative to their values at absolute zero. Generally this data appears in the form of the thermodynamic functions, $(H_T^\circ - H_0^\circ)/RT$ and $(F_T^\circ - F_0^\circ)/RT$ where the superscript ($^\circ$) denotes the quantity at standard state (the pure component at 1 atmosphere pressure). The properties H_T° and F_T° computed from these functions will hereafter be referred to as "standard" properties. In the discussion to follow we will first derive the required polynomial form for curve-fits of this data and then demonstrate the relationship necessary for obtaining the thermodynamic properties relative to the desired reference state of the elements at 298.16°K and one atmosphere.

Using the following thermodynamic relations at constant pressure:

$$dH^\circ = C_p^\circ dT \quad (1)$$

$$\frac{dS^\circ}{dT} = \frac{C_p^\circ}{T} \quad (2)$$

Standard heat capacity data can be conveniently fitted to the following polynomial form:

$$C_p^\circ = a_1 + a_2 T + a_3 T^2 + a_4 T^3 + a_5 T^4 \quad (3)$$

Substituting this relation into Equation (1) and integrating gives:

$$S_T^\circ = a_1 \ln T + a_2 T + \frac{a_3 T^2}{2} + \frac{a_4 T^3}{3} + \frac{a_5 T^4}{4} + a_7 \quad (4)$$

where a_7 is an integration constant. The use of the indefinite integral here is necessary since the polynomial formulation yields an indeterminant expression at absolute zero; however, this does not present any difficulties at temperatures other than absolute zero.

The derivative of the standard free energy of a substance can be defined in terms of standard enthalpy and entropy as

$$dF^\circ = dH^\circ - d(TS^\circ) = dH^\circ - TdS^\circ - S_T^\circ dT \quad (5)$$

From Equations (1) and (2) it is noted that

$$dH^\circ = TdS^\circ \quad (6)$$

therefore

$$dF^\circ = -S_T^\circ dT \quad (7)$$

Integrating this expression in temperature from absolute zero with S_T° defined by Equation (4) yields the following

$$F_T^\circ - F_0^\circ = \left[a_1 (\ln T - 1) T + \frac{a_2 T^2}{2} + \frac{a_3 T^3}{6} + \frac{a_4 T^4}{12} + \frac{a_5 T^5}{20} + a_7 T \right] \quad (8)$$

In general, standard free energy data is tabulated in non-dimensional form. Performing this nondimensionalization by dividing by RT and noting that $F_0^\circ = H_0^\circ$ gives

$$\frac{F_T^\circ - H_0^\circ}{RT} = A_1(1 - \ln T) - \frac{A_2}{2}T - \frac{A_3}{6}T^2 - \frac{A_4}{12}T^3 - \frac{A_5}{20}T^4 - A_7 \quad (9)$$

where $A_1 = a_1/R$, $A_2 = a_2/R$, ..., $A_5 = a_5/R$ and $A_7 = a_7/R$.

From Equations (1) and (3) the comparable polynomial expression for standard enthalpy can be derived.

$$\frac{H_T^\circ - H_0^\circ}{RT} = A_1 + \frac{A_2}{2}T + \frac{A_3}{3}T^2 + \frac{A_4}{4}T^3 + \frac{A_5}{5}T^4 \quad (10)$$

We have thus derived polynomial expressions for the thermodynamic functions of standard entropy, enthalpy and free energy relative to 0°K. In order to determine relative values of enthalpy and free energy from these functions it is necessary to specify a reference state. It is convenient to select the elements at 298°K and one atmosphere pressure for the reference state, since this condition is most widely used. The choice of reference state is related to the determination of the thermodynamic property through the enthalpy at absolute zero, H_0° . For an element, H_0° is equivalent to the change in enthalpy from the reference temperature to absolute zero. For a compound, the heat of formation from elements must be included. Since the latter quantity is identically zero for elements we can write a general expression for H_0° .

$$H_0^\circ = (\Delta H_f)_{T_{ref}} - (H_T^\circ - H_0^\circ)_{T_{ref}} \quad (11)$$

where $(H_T^\circ - H_0^\circ)_{T_{ref}}$ is available from the tabular data to be fitted.

In non-dimensional form, this equation becomes

$$\frac{H_0^\circ}{RT} = \frac{(\Delta H_f)_{T_{ref}} - (H_T^\circ - H_0^\circ)_{T_{ref}}}{RT} \quad (12)$$

Having defined a reference state it is a simple matter to determine the thermodynamic properties from the thermodynamic functions as follows:

$$\frac{F_T^\circ}{RT} = A_1(1 - \ln T) - \frac{A_2}{2}T - \frac{A_3}{6}T^2 - \frac{A_4}{12}T^3 - \frac{A_5}{20}T^4 + \frac{A_6}{T} - A_7 \quad (13)$$

$$\frac{H_T^\circ}{RT} = A_1 + \frac{A_2}{2}T + \frac{A_3}{3}T^2 + \frac{A_4}{4}T^3 + \frac{A_5}{5}T^4 + \frac{A_6}{T} \quad (14)$$

$$\text{where } A_6 = H_0^\circ / R = [(\Delta H_f)_{T_{ref}} - (H_T^\circ - H_0^\circ)_{T_{ref}}] / R \quad (15)$$

A summary of the polynomial expressions discussed in this section is given in Table 1. For the remainder of this section we will discuss the method used to obtain the appropriate coefficients for these polynomials.

Determination of Polynomial Coefficients

There are several procedures for obtaining the polynomial constants as required for the equations given in Table 1. McBride et.al.(Ref. 9), used a least squares technique which was simultaneously applied to all four of the thermodynamic functions. For the purpose of this report, the emphasis was placed upon the free energy fit rather than the properties in general. In the following paragraphs the procedure for determining these constants is explained.

From tabulated enthalpy functions as given in several reports, for example Refs. 2-9, the following polynomial was curve fit using a simple least squares analysis,

TABLE 1

A Summary of Related Polynomial Equations for Standard
Thermodynamic Properties

Specific Heat

$$\frac{C_p^\circ}{R} = A_1 + A_2 T + \frac{A_3 T^2}{2} + \frac{A_4 T^3}{3} + \frac{A_5 T^4}{4} \quad (A)$$

Enthalpy

$$\frac{H_T^\circ}{RT} = A_1 + \frac{A_2 T}{2} + \frac{A_3 T^2}{3} + \frac{A_4 T^3}{4} + \frac{A_5 T^4}{5} + \frac{A_6}{T} \quad (B)$$

Entropy

$$\frac{S_T^\circ}{R} = A_1 \ln T + A_2 T + \frac{A_3 T^2}{2} + \frac{A_4 T^3}{3} + \frac{A_5 T^4}{4} + A_7 \quad (C)$$

Free Energy

$$\frac{F_T^\circ}{RT} = A_1(1 - \ln T) - \frac{A_2 T}{2} - \frac{A_3 T^2}{6} - \frac{A_4 T^3}{12} - \frac{A_5 T^4}{20} + \frac{A_6}{T} - A_7 \quad (D)$$

$$\frac{H_T^\circ - H_0^\circ}{RT} = B_1 + B_2 T + B_3 T^2 + B_4 T^3 + B_5 T^4 \quad (16)$$

From Equation (10) the constants A_1 through A_5 were determined as shown below:

$$A_1 = B_1$$

$$A_2 = 2B_2$$

$$A_3 = 3B_3$$

$$A_4 = 4B_4$$

$$A_5 = 5B_5$$

The constant, A_6 , was computed separately from Equation (15), the appropriate values of the heat of formation 298.16 and the relative enthalpy at the reference temperature. In Appendix B the method used for calculating the heats of formation at the reference temperature is presented. Also contained in this appendix is a tabulation of the heats of formation at 298.16°K and at 0°K for each compound considered in this report.

The value of A_7 was determined as the constant difference between the tabulated free energy data and the remaining terms of the free energy polynomial as computed from the previously determined constants.

$$A_7 = A_1(1-1nT) - \frac{A_2}{2} - \frac{A_3 T^2}{6} - \frac{A_4 T^4}{12} - \frac{A_5 T^5}{20} - \left(\frac{F_T^\circ - H_0^\circ}{RT} \right) \quad (17)$$

In this report constants were evaluated at two temperature ranges for all species of interest and are listed in Appendix A. The ranges considered considered were 1000-7000°K and 5000-18000°K. The overlapping and extension of temperature ranges was necessary to overcome accuracy limitations at the extremes of the fit. For completeness, polynomial coefficients as determined by other investigators have been included for several additional species for the ranges of 300-1000°K and 1000-6000°K.

III. DISCUSSION OF RESULTS

The above procedure for determining the polynomial coefficients was implemented using a Fortran IV Computer program, a listing of which is given in Appendix C. The polynomial constants as determined by this program are given in Appendix A. As previously noted, constants from various other sources are given for additional species of interest. In Appendix D tabulations of free energy, enthalpy, heat capacity and entropy as predicted by the polynomial curvefits of this report are given for each of the species considered.

The constants which were evaluated by the method of this report give predictions of free energy and enthalpy which are accurate to a minimum of four significant figures. For most species reported, predictions are accurate to five significant figures. A random sampling of accuracy with respect to predictions of entropy and heat capacity revealed a maximum deviation of 1.58% with a mean error of 2.12%.

To further test the applicability of these results, the thermodynamic constants as determined in this study were used in several free energy minimization calculations to determine the equilibrium compositions of air and of phenolic resin-nylon composites. A comparison of these results for air at 0.68 atm. is given in Fig. 1. The thermodynamic data required for ablation products was used to compute equilibrium compositions for phenolic nylon with an elemental distribution, in mass percent, of 73.03% C, 7.29% H, 4.96% N, and 14.72% O. These results which are given in Figs. 2, 3, and 4 are in agreement with those of Stroud and Brinkley, Ref. 12. In another comparison, the equilibrium compositions of air at 1 atmosphere were then used in conjunction with predicted values of heat capacity and

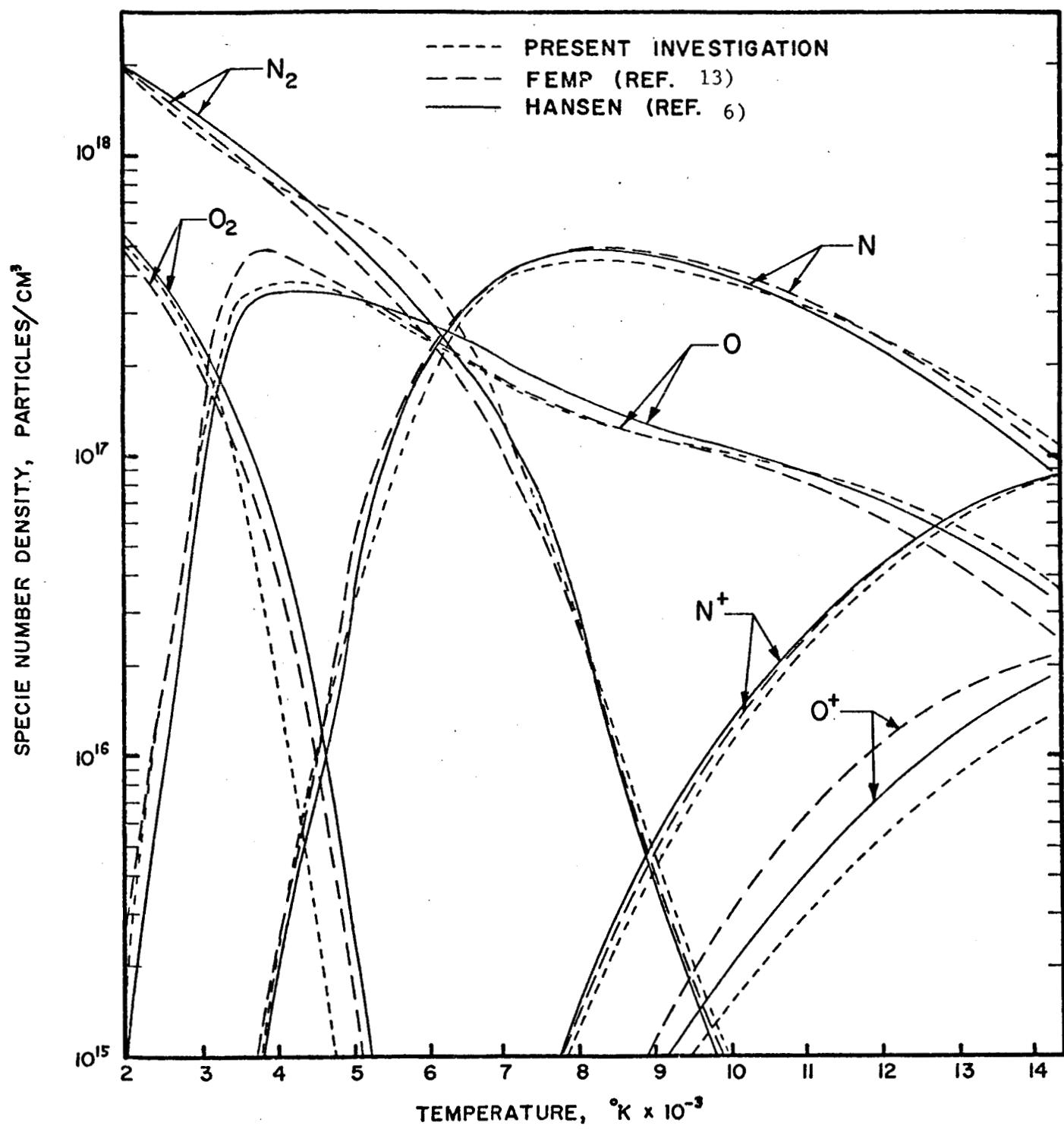


Fig. 1 Comparison of Specie Number Density Versus Temperature Calculated Using the Methods for a Pressure of .68 Atm.

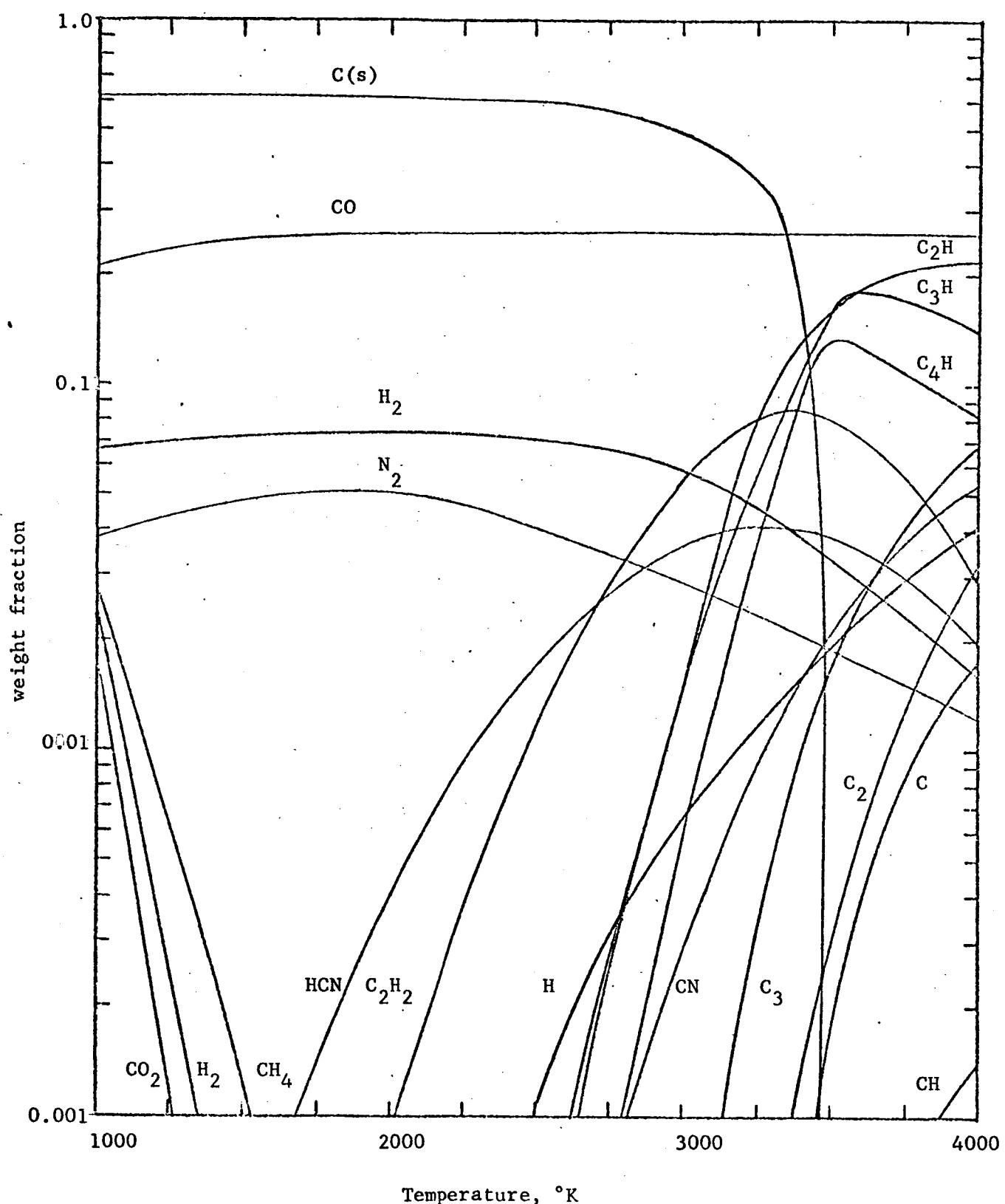


Fig. 2 Equilibrium Composition of Pyrolysis Products of Phenolic Resin-Nylon Composite at 1.0 Atmospheric Pressure

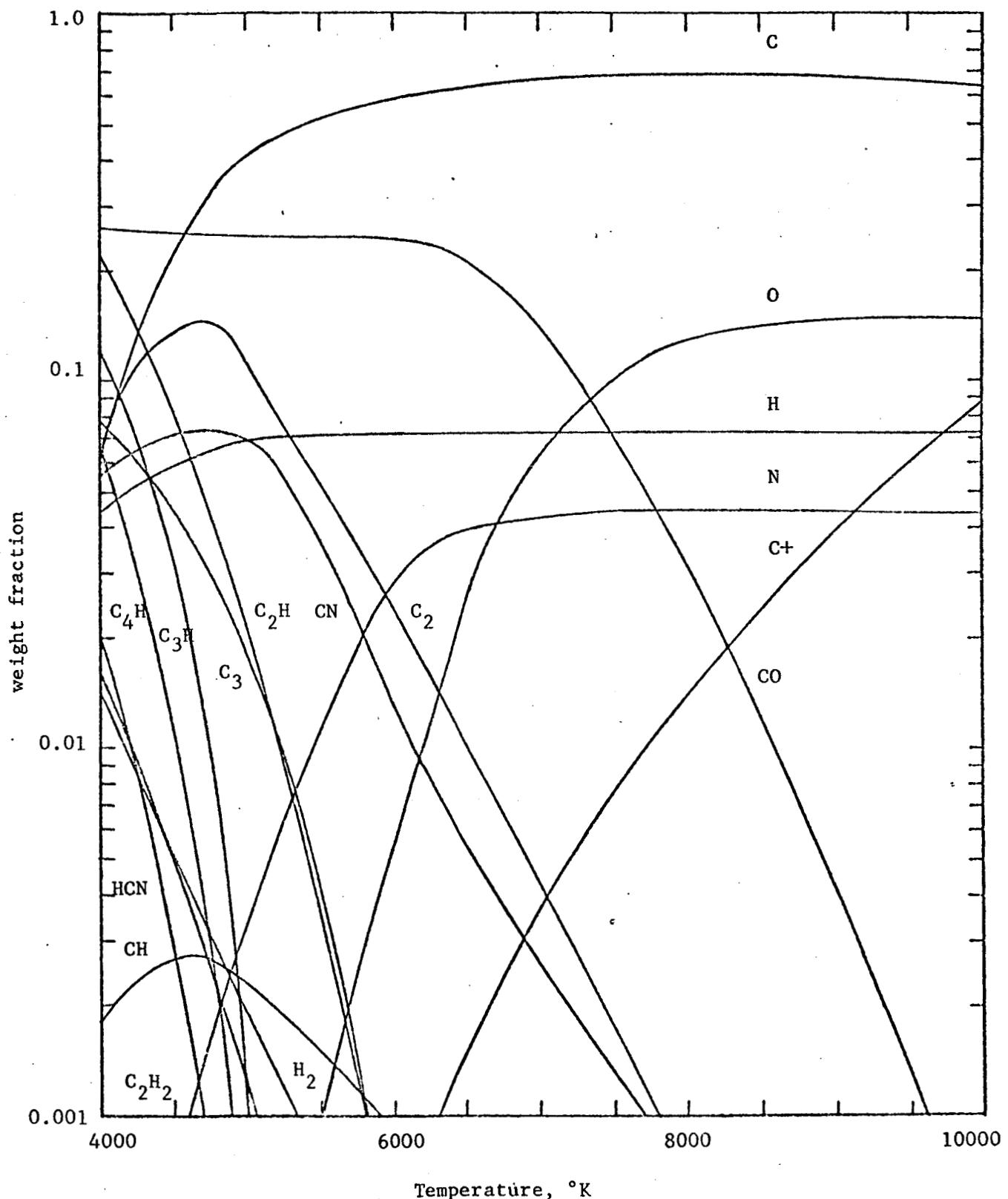


Fig. 3 Equilibrium Composition of Pyrolysis Products of Phenolic Resin-Nylon Composite at 1.0 Atmospheric Pressure

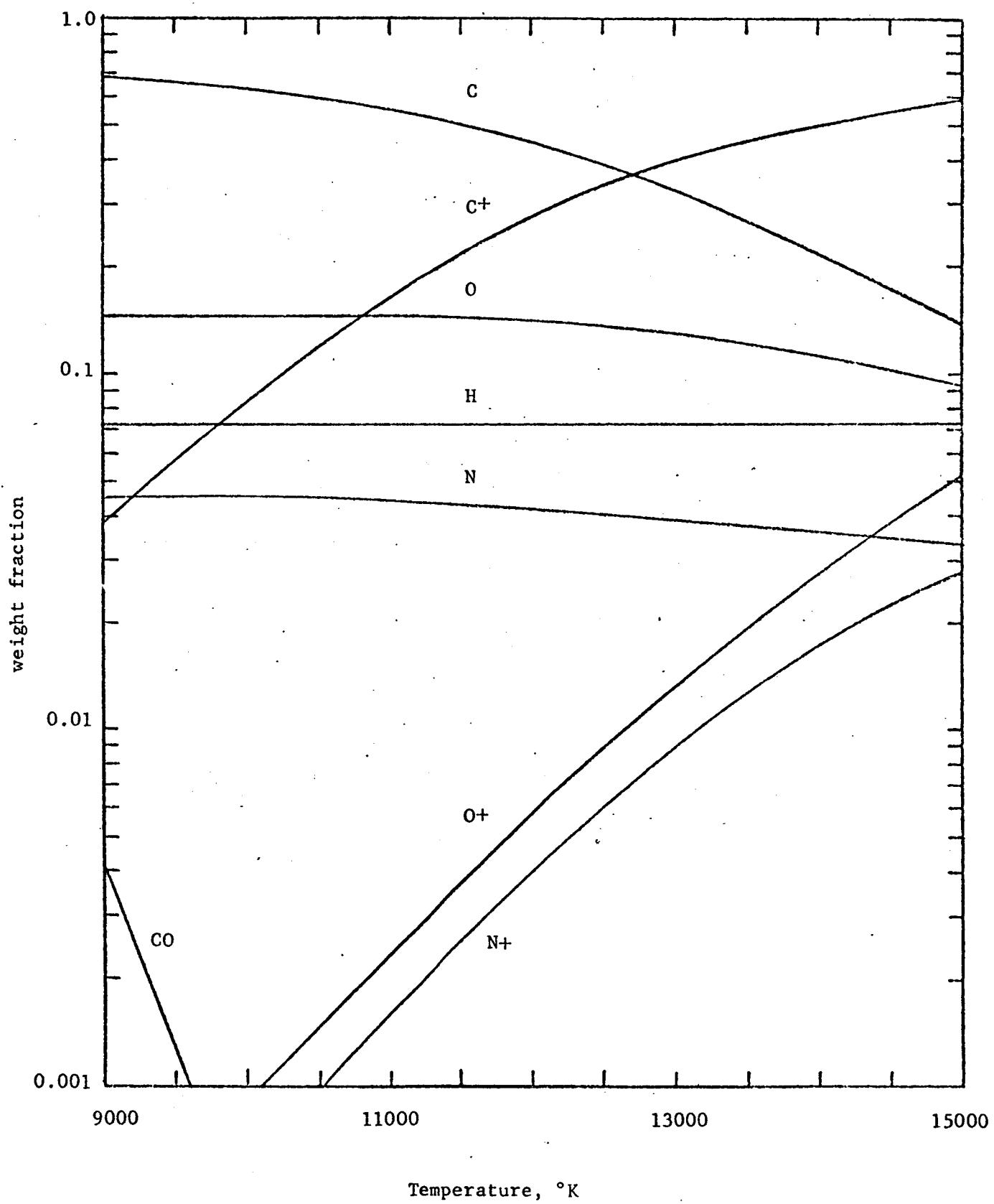


Fig. 4 Equilibrium Composition of Pyrolysis Products of Phenolic Resin-Nylon Composite at 1.0 Atmospheric Pressure

enthalpy to compute the mixture heat capacity of air by the following equation,

$$C_p^{\circ} = \sum Y_i C_{pi} + \sum H_i^{\circ} \left(\frac{\partial Y_i}{\partial T} \right) \quad (18)$$

The results were then compared with those of other investigators as shown in Figure 5. The comparison is again quite favorable.

IV. SUMMARY

The objective of this report was to construct a standard set of polynomial curve-fits of thermodynamic data for species of interest in high temperature studies of ablating thermal protection systems. In line with this objective, polynomial coefficients for temperatures from 1000-6000°K and from 6000-15000°K were determined for 53 chemical species using the method described in Section II.

Making use of compatible curve-fits from other sources, this list was extended to 99 species, many having constants reported for three temperature ranges from 300-15000°K. Constants for the remaining species are reported either for the 300-1000°K and 1000-6000°K ranges or the 1000-6000°K and 6000-15000°K ranges. It should be noted that those species whose coefficients are not reported for a particular temperature range, are not likely to exist in equilibrium within the omitted range.

Using the computer program given in Appendix C, similar curve-fits can be obtained for any given component from tabulations of its free energy and enthalpy functions and the heat of formation of the desired component. Thus as a result of this report, a standard set of polynomial curve-fits is now available for all species of interest for studies of ablating thermal protection systems.

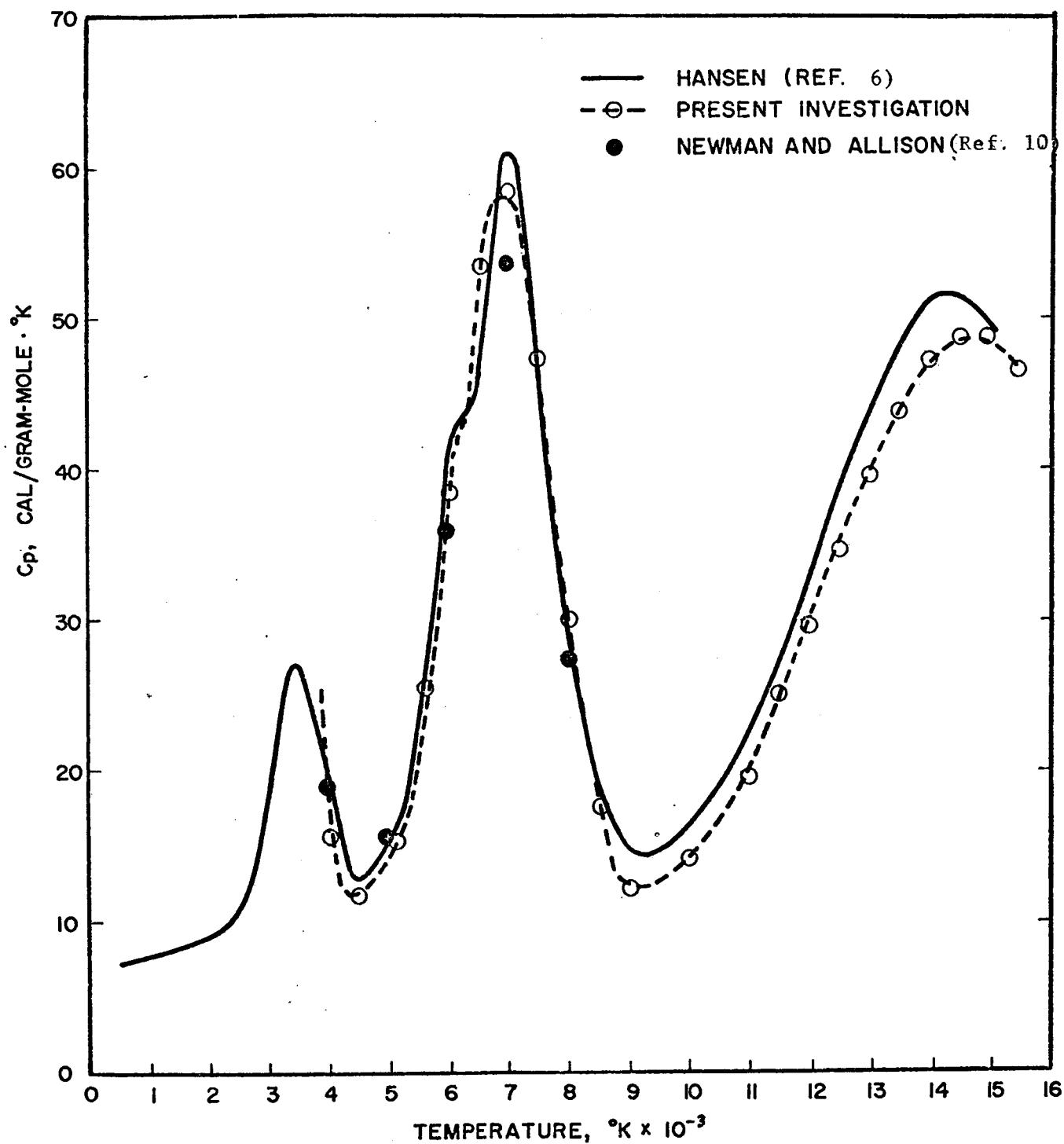


Fig. 5 Comparison of Mixture Reacting Heat Capacity for Air at 1 Atm.

V. BIBLIOGRAPHY

1. Browne, W. G., "Thermodynamic Properties of Some Atoms and Atomic Ions", General Electric Company MSVD Engineering Physics Technical Memorandum #2.
2. Browne, W. G., "Thermodynamic Properties of Some Diatoms and Diatomic Ions," General Electric Company MSVD Engineering Physics Technical Memorandum #8, (May 14, 1962).
3. Browne, W. G., "Thermodynamic Properties of the Species CN, C₂, C₃, C₂N₂ and C", General Electric Company MSVD Engineering Physics Technical Memorandum #9, (May 14, 1962).
4. Browne, W. G., "Thermodynamic Properties of Some Ablation Products from Plastic Heat Shield in Air", General Electric Company MSVD Engineering Physics Technical Memorandum #11, (March 15, 1964).
- 4-A. Duff, Russell E. and S. H. Bauer, "Equilibrium Composition of the C/H System at Elevated Temperatures", Jour. Chem. Physics, 36, 1754 (1962).
5. Gurvich, A. B., G. A. Khachkurov, et al., "Thermodynamic Properties of Individual Substance. Calculation of the Thermodynamic Properties", Vol. 1-3, Foreign Technology Division Wright Patterson Air Force Base, Ohio, AD 659 659, AD 659 660 and AD 659 679 (May, 1967)
6. Hansen, C. F., "Approximations for the Thermodynamic and Transport Properties of High-Temperature Air", NASA TR R-50, (1959).
7. Huff, Vearl N., Sandord Gordon, and Virginia E. Morrel, NASA TR 1037 (1951).
8. Marrone, P. W., "Normal Shock Waves in Air Equilibrium Composition and Flow Parameters for Velocities from 26,000 to 50,000 ft/sec." Cornell Aeronautical Laboratory Inc., CAL Report No. AG-1729-A-2, Contract No. NASr-119 (August, 1962).
9. McBride, B. J., S. Heimel, J. G. Ehler, S. Gordon, "Thermodynamic Properties to 6000°K for 210 Substances Involving the First Eighteen Elements", NASA SP-3001 (1963).
10. Newman, P. A. and D. O. Allison, "Direct Calculations of Specific Heats and Related Thermodynamic Properties of Arbitrary Gas Mixtures With Tabulated Results", NASA TN D-3540, (1966).
11. Stull, D. R., et al., JANAF Thermochemical Tables. The Dow Chemical Co., Midland, Mich., Dec. 31, 1960-Sept. 30, 1962.

12. Stroud, C. W. and K. L. Brinkley, "Chemical Equilibrium of Ablation Materials Including Condensed Species", NASA TN-D-5391 (August 1969).
13. Browne, H. N., M. W. Williams and D. R. Cruise, "Theoretical Computation of Equilibrium Compositions, Thermodynamic Properties and Performance Characteristics of Propellant Systems", U. S. Naval Ordnance Test Station NAVWEPS, Rept. 7043, June 1960.

APPENDIX A

TABULATED CONSTANTS FOR POLYNOMIAL APPROXIMATIONS
OF THERMODYNAMIC PROPERTIES

This appendix contains in tabular form the necessary constants for prediction of enthalpy, free energy, heat capacity and entropy, by the following equations.

$$\frac{H_T^\circ}{RT} = A_1 + \frac{A_2}{2}T + \frac{A_3}{3}T^2 + \frac{A_4}{4}T^3 + \frac{A_5}{5}T^4 + \frac{A_6}{6}$$

$$\frac{F_T^\circ}{RT} = A_1(1 - \ln T) - \frac{A_2}{2}T - \frac{A_3}{6}T^2 - \frac{A_4}{12}T^3 - \frac{A_5}{20}T^4 + \frac{A_6}{T} - A_7$$

$$\frac{C_P^\circ}{R} = A_1 + A_2 T + A_3 T^2 + A_4 T^3 + A_5 T^4$$

$$\frac{S_T^\circ}{R} = A_1 \ln T + A_2 T + \frac{A_3}{2}T^2 + \frac{A_4}{3}T^3 + \frac{A_5}{4}T^4 + A_7$$

The constants, A_1 through A_7 , are given in Table A-1 where the following notation is used:

SPECIE	pure component
CODE	the reference source of data
B1, B2, B3	data source from Browne (2,3,4,5) for temperature regions 300°K to 1000°K, 1000°K to 6000°K and 6000°K to 15000°K respectively.
D1, D2	data source from Duff (4A), for temperature regions 300°K to 1000°K and 6000°K to 15000°K respectively
M1, M2	data source from McBride (9), for temperature regions 300°K to 1000°K and 1000°K to 6000°K respectively
*	constants computed in the reference shown
A1-A7	the constants for the thermodynamic functions

NOT REPRODUCIBLE

A+	0.2529E 01	0.3151E-03	-0.1961E-06	0.3988E-10	-0.2618E-14	0.1828E 06	0.5481E 01	B2
	0.2741E 01	-0.9153E-04	0.1409E-07	-0.9643E-12	0.2416E-16	0.1828E 06	C.4416E C1	B3
A++	0.2702E 01	0.1121E-03	-0.1210E-06	0.3426E-10	-0.2591E-14	0.5041E 06	0.4881E 01	32
	0.2423E 01	0.1366E-03	-0.2538E-08	-0.8412E-12	C.354CE-16	C.5041E 06	C.6704E 01	B3
A+++	0.2473E 01	0.7329E-04	-0.5893E-07	0.1551E-10	-0.9554E-15	0.9795E 06	0.5855E 01	82
	0.2866E 01	-0.4335E-03	0.1226E-06	-0.9547E-11	0.2329E-15	0.9795E 06	0.3669E 01	B3
A-	C.2500E 01	0.1016E-05	-0.6260E-09	0.1346E-12	-0.9238E-17	-0.7454E C3	0.4367E 01	B2
	C.3079E 01	-0.5709E-03	0.1538E-06	-0.1596E-10	0.5712E-15	-0.7454E 03	C.9579E 00	93
C+	0.2659E 01	-9.1393E-03	0.5959E-07	-0.1037E-10	0.6345E-15	C.2168E C6	0.3709E 01	32
	0.2528E 01	0.4869E-05	-0.7026E-08	0.1134E-11	-C.3476E-16	0.2168E 03	0.4139E 01	33
C++	0.2501E 01	-0.4142E-05	0.3365E-08	-C.9705E-12	C.9097E-16	0.5004E 06	0.2558E 01	B2
	0.2304E 01	0.2051E-03	-0.5698E-07	0.5657E-11	-C.154CE-15	0.5004E 06	0.3709E 01	B3
C+++	C.2500E 01	-0.1301E-06	0.1364E-09	-0.4498E-13	C.4405E-17	0.1057E 07	0.3257E 01	B2
	C.2460E 01	0.2078E-04	-0.7221E-08	0.591CE-12	-C.1099E-16	C.1057F 07	C.3444E 01	B3
C(S)	-0.7124E 00	0.7341E-C2	-0.5526E-C5	0.1514E-08	-0.2382E-13	-0.6805E C2	0.0	*M1
	0.7969E-02	C.4899E-02	-0.2569E-C5	0.6697E-09	-C.6466E-13	-C.1264E 03	-0.5150E 00	*M2
C	0.2541E 01	-0.2210E-03	0.4736E-06	-0.4529E-09	0.1602E-12	0.8545E 05	0.4573E 01	*M1
	0.2612E 01	-0.2030E-03	0.1095E-06	-0.1695E-10	C.8590E-15	0.8542E 05	0.4144E 01	B2
	0.2141E 01	0.3219E-03	-0.5498E-07	C.3604E-11	-0.5564E-16	0.3542E 05	0.6874E C1	B3
C-	C.2500E 01	0.344CE-06	-0.1954E-09	0.3937E-13	-0.2573E-17	0.7C83E 05	0.4356E 01	B2
	C.2508E 01	-0.6332E-05	0.1364E-08	-0.1094E-12	C.2934E-17	0.7083E 05	0.4309E 01	B3

C14	C.3545E 01 C.3186E 01 C.3963E 01	0.3795E-04 0.9461E-03 0.1127E-03	-0.1826E-05 -0.2442E-06 0.2193E-07	0.4455E-08 0.3107E-10 -0.1693E-11	-0.2186E-11 -0.1353E-14 0.2844E-16	0.7060E 05 0.7061E 05 0.7061E 05	0.1824E 01 0.3504E 01 -0.6141E 00	*M1 B2 B3
CH2	C.3551E 01 C.3274E 01 C.4555E 01	-0.2507E-02 0.2638E-02 0.9757E-03	0.1235E-04 -0.7538E-06 -0.1478E-06	-0.1175E-07 0.9695E-10 0.9828E-11	0.3812E-11 0.4625E-14 -0.2402E-15	C.3366E 05 0.3365E 05 0.3365E 05	0.1797E 01 0.3183E 01 -0.4053E 01	*M1 B2 B3
CH3	C.3406E 01 C.3529E 01 C.6151E 01	0.4268E-02 0.5214E-02 0.1563E-02	0.2033E-06 -0.1680E-05 -0.2364E-06	-0.1155E-03 0.2399E-09 0.1576E-10	0.4129E-12 -0.1251E-13 -0.3859E-15	0.1565E 05 C.1472E 05 0.1472E 05	0.2704E 01 0.1893E 01 -0.1272E 02	*M1 B2 B3
CH4	C.4250E 01 C.2234F 01 C.6610E 01	-0.6913E-02 0.8929E-02 0.2665E-02	0.3160E-04 -0.2940F-05 -0.4192E-06	-0.2971E-07 0.4264E-09 C.2880F-10	0.9510E-11 -0.2248E-13 -0.7251E-15	-C.1019E 05 -C.1C21E 05 -C.1021E 05	-0.9175E 00 0.6764E 01 -0.1751E C2	*M1 B2 B3
CN	C.3353E 01 C.3411E 01 C.3473E 01	-0.2763E-02 0.4897E-03 0.7337E-03	0.6357E-05 0.1050E-06 -0.9088E-07	-0.5413E-09 -0.3473E-10 0.4847E-11	C.1491E-11 C.2361E-14 -0.1018E-15	0.4741E 05 0.4745E 05 0.5420E 05	0.2972E 01 0.4746E 01 0.4152E 01	*M1 B2 B3
CD+	C.3148E 01 C.4076E 01	0.1265E-02 -0.1496E-03	-0.4715E-06 0.1044E-06	0.7965E-10 -0.9201E-11	-0.4421E-14 C.2310E-15	0.1490E 06 0.1490E 06	0.6051E 01 0.9992E 00	B2 B3
CO	C.3787E 01 C.3254E 01 C.3366E 01	-0.2171E-02 0.9698E-03 0.8027E-03	0.5076E-05 -0.3474E-08 -0.2647E-06	-0.3474E-08 0.3037E-10 0.1968E-06	C.7722E-12 -C.1177E-14 -0.5549E-15	-0.1436E 05 -C.1434E 05 -0.1434E 05	0.2634E 01 0.4875E 01 0.4263E 01	*M1 B2 B3
C02	C.2173E 01 C.4413E 01	0.1038E-01 0.3192E-02	-0.1073E-04 -0.1298E-05	0.6346E-08 0.2415E-09	-0.1620E-11 -0.1674E-13	-0.4835E 05 -0.4894E 05	0.1066E 02 -0.7288E 00	*M1 M2

NOT REPRODUCIBLE

CJS	0.2939E-01	0.1461E-01	-0.2046E-04	0.1566E-07	-0.4446E-11	-0.1762E-05	0.1237E-02	*M1
	0.5237E-01	0.2417E-02	-0.1001E-05	0.1879E-09	-0.1310E-13	-0.1833E-05	-0.2913E-01	*M2
CS	0.3398E-01	-0.5892E-03	0.5945E-05	-0.7086E-08	0.2616E-11	0.2648E-05	0.5919E-01	*M1
	0.3677E-01	0.9269E-03	-0.3887E-06	0.7439E-10	-0.5249E-14	0.2692E-05	0.3916E-01	*M2
CS2	0.2917E-01	0.1250E-01	-0.1611E-04	0.1057E-07	-0.2794E-11	0.1278E-05	0.8876E-01	*M1
	0.5949E-01	0.1724E-02	-0.7211E-06	0.1374E-09	-0.9634E-14	0.1205E-05	-0.6205E-01	*M2
C2	0.7509E-01	-0.1065E-01	0.1008E-04	-0.8621E-09	-0.1752E-11	0.9880E-05	-0.1603E-02	*M1
	0.4442E-01	-0.2835E-03	0.3036E-06	-0.6244E-10	0.3915E-14	0.9787E-05	-0.1090E-01	B2
	0.4526E-01	0.4857E-03	-0.7026E-07	0.4666E-11	-0.1142E-15	0.4787E-05	0.1090E-01	C1
	0.3485E-01	0.3563E-02	-0.1237E-05	0.1866E-09	-0.1013E-13	0.5809E-05	0.4784E-01	B2
	0.5307E-01	0.8966E-03	-0.1378E-06	0.9251E-11	-0.2278E-15	0.5809E-05	-0.5283E-01	B3
C2H2	0.7903E-00	0.2347E-01	-0.3554E-04	0.2795E-07	-6.8448E-11	0.2625E-05	0.1401E-02	*M1
	0.3891E-01	0.5717E-02	-0.1957E-05	0.2931E-09	-0.1585E-13	0.2590E-05	0.6520E-00	B2
	0.6789E-01	0.1563E-02	-0.2295E-06	0.1534E-10	-0.3763E-15	0.2590E-05	-0.1539E-02	B3
C2H3	0.3486E-01	0.8253E-02	-0.2820E-05	0.4209E-09	-0.2268E-13	0.3140E-05	0.5278E-01	B2
	0.7538E-01	0.2298E-02	-0.3642E-06	0.2519E-10	-0.6378E-15	0.3140E-05	-0.1709E-02	B3
C2H4	0.1125E-01	0.1391E-01	0.2657E-05	-0.1156E-07	0.5239E-11	0.5333E-04	0.1584E-02	*M1
	0.2754E-01	0.1172E-01	-0.4053E-05	0.6098E-09	-0.3303E-13	0.6026E-04	0.7185E-01	B2
	0.3627E-01	0.3039E-02	-0.4864E-06	0.3342E-10	-0.8406E-15	0.6026E-04	-0.2516E-02	B3
C2H6	0.2382E-01	0.1032E-01	0.1191E-04	-0.1483E-07	0.4475E-11	-0.1162E-05	0.7598E-01	*D1
	0.1431E-01	0.1889E-01	-0.7044E-05	0.1187E-08	-0.7445E-13	-0.1143E-05	0.1401E-02	*D2

C2H2	0.3403E 01 0.5396E 01 -C.2697E 01	0.1776E-01 0.5089E-02 0.1308E-01	-0.2686E-C4 -0.1880E-C5 -0.2887E-05	C.2197E-07 0.2931E-C9 0.4576E-09	-C.7987E-11 -0.162CE-13 -0.1878E-13	0.3555E 05 0.3563E 05 0.3563E 05	0.5412E 01 -0.2598E 01 0.4498E 02	*D1 *D2 *D3
C3	0.2633E C1 0.4002E C1 0.2213E C2	0.9419E-02 0.3541E-02 -0.1759E-01	-0.9593E-05 -0.1318E-05 0.5565E-05	0.5580E-08 0.2064E-09 -0.6758E-09	-0.1424F-11 -0.1144E-13 0.2825E-13	C.9431E 05 0.9423E 05 0.9423E 05	0.8079E 01 0.2020E 01 -0.1021E 03	*D1 *D2 *D3
C3H	0.2+74E C1	0.1175E-01	-0.8045E-05	0.2729E-08	-C.3544E-12	C.6308E 05	0.1054E 02	*D1
C3H2	0.1568E C1 0.4552E C1	0.2219E-01 0.8696E-02	-0.2193E-04 -0.3154E-05	0.1110E-07 0.5242E-09	-0.2153E-11 -0.3264E-13	0.5230E 05 0.5191E 05	0.1304E 02 0.1827E 00	*D1 *D2
C3H3	0.2588E C1 0.4556E C1	0.1894E-01 0.1051E-01	-0.1393E-04 -0.3955E-05	0.5541E-08 0.6723E-09	-0.9375E-12 -0.4243E-13	C.3700E C5 0.3667E 05	0.1032E 02 0.6294E 00	*D1 *D2
C3H5	0.3159E C1 0.4153E C1	0.1511E-C1 0.1671E-C1	0.44G7E-C5 -0.6328E-C5	-0.9188E-C8 0.1080E-08	C.2598E-11 -0.6880E-13	C.1469E 05 C.1439E 05	0.1374E 02 0.7926E 01	*D1 *D2
C3H2	0.2056E C1 0.1291E C1	0.2528E-01 0.2775E-C1	-0.5914E-C6 -0.1047E-C4	-0.4880E-C8 C.1761E-C8	0.3199E-11 -0.1124E-12	-C.1426E 05 -C.1407E 05	0.1318E 02 0.1715E 02	*D1 *D2
C4	0.5670E C1 0.6077E C1	0.5383E-02 0.3444E-02	-0.4751E-C5 -0.9480E-C6	C.2349E-08 0.9346E-10	-0.4548E-12 -0.1791E-14	0.1195E 06 0.1195E 06	-0.4253E 01 -0.5995E 01	*D1 *D2
C4H1	0.2696E C1	0.2268E-01	-0.2451E-C4	0.1315E-07	-0.2673E-11	0.7649E C5	C.1040E 02	*D1
C4H2	0.6216E C1 0.1118E C2	0.9660E-C2 C.2194E-C2	-0.3489E-C5 -C.3305E-C6	0.5411E-09 C.2292E-10	-0.2996E-13 -0.5670E-15	0.5605E 05 0.5605E 05	-0.1211E 02 -C.3943E C2	B2 B3

NOT REPRODUCIBLE

C4H3	0.5749E 01	0.1248E-01	-0.4751E-05	0.8136E-09	-0.5170E-13	0.4910E 05	-0.3430E 01	*D2
C5	0.7279E 01	0.6762E-02	-0.5026E-05	0.2532E-08	-0.5450E-12	0.1186E 06	-0.1311E 02	*D1
	0.7577E 01	0.4618E-02	-0.1271E-05	0.1188E-09	-0.1123E-14	0.1186E 06	-0.1431E 02	*D2
C5H	0.1022E 01	0.3109E-01	-0.3043E-04	0.1481E-07	-0.2783E-11	0.9217E 05	0.1757E 02	*D1
	0.5465E 01	0.1118E-01	-0.4300E-05	0.7241E-09	-0.4441E-13	0.9152E 05	-0.2844E 01	*D2
C5H2	0.1006E 01	0.4108E-01	-0.4555E-04	0.2475E-07	-0.5060E-11	0.8139E 05	0.1669E 02	*U1
	0.6655E 01	0.1325E-01	-0.5164E-05	0.8323E-09	-0.5345E-13	0.8063E 05	-0.8785E 01	*D2
C5H3	0.2649E C1	0.3462E-C1	-0.3266E-C4	0.1602E-07	-0.3038E-11	0.6608E 05	0.1140E 02	*U1
	0.7099E C1	0.1455E-C1	-0.5664E-C5	0.9844E-09	-0.6320E-13	0.6544E 05	-0.9039E 01	*D2
C5H6	0.8735E 00	0.2770E-01	0.4256E-05	-0.1440E-07	0.4466E-11	0.8995E 04	0.1905E 02	*D1
	C.3601E 01	0.2669E-01	-0.1637E-04	0.1800E-03	-0.1155E-12	0.8326E 04	0.4317E 01	*D2
C6	0.8888E 01	0.7644E-02	-0.5307E-C5	0.2779E-08	-0.6265E-12	0.1437E 06	-0.1837E 02	*D1
	C.9077E 01	0.5792E-02	-0.1595E-05	0.1441E-09	-0.4528E-15	0.1437E 06	-0.1904E 02	*D2
C6H	0.2717E C1	0.3941E-01	-0.4505E-C4	0.2477E-07	-0.5100E-11	0.1051E 06	0.1184E 02	*D1
	0.8362F 01	0.1165E-01	-0.4637E-C5	0.8085E-C9	-0.5150E-13	C.1043E 06	-0.1363E 02	*D2
C6H2	0.2324E 01	0.4630E-01	-0.5461E-C4	0.3087E-07	-0.6480E-11	0.9307E 05	0.1082E 02	*D1
	0.8775E 01	0.1341E-C1	-0.5232E-05	0.9113E-09	-0.5860E-13	0.8223E 05	-0.18C5E 02	*D2
C6H3	0.2926E C1	0.4065E-01	-0.3887E-C4	0.1907E-07	-0.3666E-11	C.7738E 05	0.1118E 02	*D1
	0.8313E 01	0.164CE-01	-0.6416E-C5	0.1119E-08	-0.7200E-13	0.7660E 05	-0.1357E 02	*D2
C6H6	0.3511E 00	0.3754E-01	-0.7947E-C5	-0.7502E-08	0.3038E-11	0.8204E 04	0.1987E 02	*D1
	C.4620E 01	0.2881E-01	-0.1124E-04	0.1958E-08	-0.1257E-12	0.7392E 04	-0.1789E 01	*D2

C7	0.105CE 0.1058E	C2 0.2	0.8524E-02 0.6953E-02	-0.5585E-05 -0.1917E-05	0.2994E-08 0.1689E-09	-0.7075E-12 0.2555E-15	0.1436E 0.1436E	0.1436E -0.2736E	0.2722E 0.2	*D1 *M2
C7H	0.1744E 0.9324E	C1 0.1	0.4949E-01 0.1425E-01	-0.5613E-04 -0.5734E-05	0.3040E-07 0.1009E-08	-0.6175E-11 -0.6475E-13	0.1194E 0.1184E	0.1562E -0.1734E	0.2 0.2	*D1 *D2
C7H2	C.2630E C.9139E	C1 C1	C.4776E-01 C.1758E-01	-0.5111E-04 -0.6763E-05	0.2777E-07 0.1167E-08	-0.5745E-11 -0.7450E-13	0.1089E 0.1089E	C.1369E -0.1389E	0.2 0.2	*D1 *D2
C8	0.1211E C.1204E	C2 C2	0.9404E-02 0.8134E-02	-0.5861E-05 -0.2233E-05	0.3208E-08 0.1935E-09	-0.7885E-12 0.9902E-15	0.1696E 0.1696E	0.3249E -0.3210E	0.2 0.2	*D1 *D2
C8H	C.1532E C.9773E	C1 C1	C.5652E-01 C.1653E-01	-0.6394E-04 -0.8198E-05	0.3473E-07 0.1181E-08	-0.7095E-11 -0.7509E-13	0.4291E 0.1418E	C.6400E-01 -0.2084E	0.2 0.2	*D1 *D2
C8H2	0.1087E C.1047E	C1 C2	0.6487E-01 0.1820E-C1	-0.7572E-04 -0.7243E-05	0.4267E-07 0.1278E-08	-0.8709E-11 -0.8285E-13	0.1112E 0.110CE	C.1567E -0.2657E	0.2 0.2	*D1 *D2
C9	0.1371E 0.1358E	C2 C2	0.1C29E-01 0.9308E-02	-0.6144E-05 -0.2562E-05	0.3427E-08 0.2189E-09	-0.8705E-12 0.1542E-14	0.1670E 0.1670E	C.4134E -C.4041E	0.2 0.2	*D1 *D2
C9H	C.1099E C.1114E	C1 C2	C.6769E-C1 C.1843E-01	-0.7887E-04 -0.7453E-05	0.4345E-07 0.1318E-C8	-0.8925E-11 -0.651CE-13	0.1448E 0.1434E	C.1813E -C.2008E	0.2 0.2	*D1 *D2
C9H2	0.1692E C.1066E	C1 C2	0.6336E-01 0.2136E-C1	-0.7042E-04 -0.8502E-05	0.3634E-07 0.1477E-08	-C.7975E-11 -0.9480E-13	0.1342E C.1331E	0.1499E -0.2275E	0.2 0.2	*D1 *D2
C10	0.1532E C.1530E	C2 C2	0.1117E-01 0.1048E-01	-0.6422E-C5 -0.2865E-C5	0.3642E-C8 0.2439E-09	-0.9510E-12 0.2339E-14	0.1965E 0.1966E	-0.4661E -0.4514E	0.2 0.2	*D1 *D2

C1044	0.2244E C1 0.1255E C2	0.7022E-01 -0.8037E-04 0.2911E-02 -0.8136E-05	0.4395E-07 -0.9005E-11 0.1440E-08 -0.9305E-13	0.1714E 06 0.1708E 06	0.1442E 02 0.2678E 02	*D1 *U2
C1042	0.1810E C1 C.1322E C2	0.7857E-01 -0.245E-04 C.2165E-01 -0.866CE-05	0.5162E-07 -0.1072E-10 0.1532E-08 -0.9955E-13	0.1396E 06 0.1381E 06	0.1360E 02 0.3773E 02	*D1 *D2
E-	0.2500E C1 0.2504E C1	0.344CE-06 -0.1954E-09 -0.6332E-05 0.1364E-08	0.3937E-13 -0.2573E-17 -0.1994E-12 0.2934E-17	-0.745CE 03 -0.7450E 03	-0.1173E 02 -0.1208E 02	32 33
H	0.250CE C1 C.2500E C1 C.3934E C1	0.C.0 -0.8243E-05 -0.1776E-02	0.0 0.6421E-C9 0.6013E-06	0.0 -0.1720E-12 -0.7819E-10	0.2547E 05 0.2547E 05	-0.460CE 00 -0.4612E 00
HCN	0.2198E C1 0.3654E C1 0.3654E C1	0.1073E-01 -0.1509E-04 0.3444E-02 -0.1258E-05 0.3444E-02 -0.1258E-05	0.1193E-07 -0.370CE-11 0.2169E-09 -0.1430E-13 0.2169E-09 -0.1430E-13	0.1468E 05 0.1442E 05 0.1442E 05	0.9281E 01 0.2373E 01 0.2373E 01	*M1 B2 *M3
HC0+	0.3463E C1 C.5093E C1	0.3155E-02 -0.1091E-05 0.7856E-03 -0.1212E-06	0.1636E-09 -0.8836E-14 0.8161E-11 -0.2013E-15	0.1027E 06 0.1027E 06	0.6279E 01 -0.2739E 01	B2 B3
HC0	C.3865E C1 C.3503E C1 C.5092E C1	-0.5371E-03 0.6903E-05 0.3025E-02 -0.1004E-05 0.7856E-03 -0.1212E-06	-0.6645E-08 C.1444E-09 0.8161E-11	-0.2658E-11 -0.7512F-14 -0.2013E-15	-0.2800E 04 -0.2824F 04 -0.2824E 04	0.4897E 01 C.6061F 01 -0.2739E 01
HU2	C.3632E C1 0.5071E C1	0.2849E-02 -0.9587E-06 0.7800E-03 -0.1191E-06	0.1417E-09 -0.7590E-14 0.7957E-11 -0.1952E-15	0.1326E 04 0.1326E 04	C.5119E 01 -0.2862E 01	B2 B3
HS	C.4259E C1 C.2982E C1	-0.1278E-02 -0.8379E-06 0.1359E-02 -0.4711E-06	0.3832E-08 -0.2025E-11 0.7904E-10 -0.5058E-14	0.1702E 05 0.1739E 05	-0.3537E 00 0.6334E 01	*M1 *M2

H2	C.2849E 01	0.4193E-02	-0.9612E-05	C.9512E-08	-0.3309E-11	-0.9673E C3	-0.1412E 01	*M1
	C.3353E C1	C.2794E-03	0.9372E-07	-C.2448E-10	C.2141E-14	-0.1C18E C4	-0.3548E 01	B2
	C.3563E 01	0.4656E-03	-0.5127E-07	0.2302E-11	-C.4905E-16	-0.1018E C4	-0.3716E 01	B3
H2CO	C.2902F C1	0.6323E-02	-0.2188E-05	0.3284E-09	-0.1775E-13	-0.2605E 04	0.7722E C1	32
	C.6181E C1	0.1566E-02	-0.2407E-06	C.1614E-10	-C.3966E-15	-0.2605E 04	-0.1039E 02	B3
H2O	C.4156E C1	-0.1724E-02	0.5693E-05	-0.4593E-08	C.1423E-11	-0.3029E 05	-0.6862E C0	*M1
	C.3387E C1	C.2140E-02	-0.5C48E-06	0.5269E-10	-0.1999E-14	-0.3028E C5	0.2635E 01	B2
	C.4339E 01	0.1339E-02	-0.1545E-06	C.1017E-10	-0.2468E-15	-0.3028E C5	-0.2837E 01	B3
H3O+	C.2934E 01	0.4980E-02	-0.1439E-05	0.1882E-C9	-C.9155E-14	C.6955E 05	0.4949E C1	32
	C.5321F C1	0.1848E-02	-0.2780E-C6	0.1839E-10	-C.4476E-15	0.6955E 05	-0.8506E C1	A3
H2S	C.3916F C1	-0.3514E-03	0.4219E-05	-0.2745E-08	C.4858E-12	-0.3609E 04	0.2366E' C1	*M1
	C.2766E C1	0.4013F-02	-0.1564E-05	C.2631E-09	-0.1797E-13	-C.3386E C4	C.7932E C1	*H2
N+	C.2727E 01	-0.2826E-03	C.1105E-06	-0.1551E-10	0.7847E-15	0.2254E 06	0.3645E C1	32
	C.2499E 01	-0.3725E-C5	C.1147E-C7	-C.11C2E-11	0.3078E-16	0.2254E 06	0.4950E 01	B3
N++	C.2775E C1	-0.3335E-03	0.1403E-06	-0.2374E-10	C.1387E-14	0.4557E C6	0.2946E 01	32
	C.2559E 01	-0.6476E-05	-0.2512E-C8	0.3786E-12	-0.3326E-17	0.4557E C6	C.4136E 01	B3
N+++	C.2500E C1	-0.4722E-06	C.3938E-C9	-0.1124E-12	C.1016E-16	0.1C08E C7	0.2794E C1	32
	C.2464F C1	0.8766E-04	-0.2C49E-C7	0.1624E-11	-C.3903E-16	C.1008E C7	0.3369E 01	B3
N	C.2515E C1	-0.1124E-03	0.2965E-06	-0.3240E-09	0.1259E-12	0.5613F C5	0.4119E 01	*M1
	C.2474E 01	0.9097E-04	-0.7814E-C7	C.2218E-10	-C.1489E-14	C.5609E 05	C.43C0E 01	B2
	C.2746F 01	-0.3909E-03	C.1238E-Q6	-C.1191E-10	C.3369E-15	C.5609E 05	C.2872E 01	B3

NH	0.3455E 01 0.2727F C1	0.528CE-03 -0.1973E-05 0.142CE-02 -0.4583E-06	0.2958E-08 -0.1208E-11 0.7555E-10 -0.4617E-14	0.3867E 05 0.3893E C5	0.2010E 01 0.6044E C1	*M1 *N2
NH3	0.3772E C1 0.2149E C1	-C.4862E-03 0.9874E-05 C.6493E-02 -0.2269E-05	-0.9568E-08 0.3131E-11 C.3739E-C9 -C.2361E-13	-0.6728E 04 -0.6401E C4	0.1465E 01 0.9820E 05	*M1 *N2
NO+	0.3200E 01 0.3561E C1	0.1029E-02 -0.3075E-06 0.6028E-03 -0.1540E-06	0.4028E-10 -0.1814E-14 0.1697E-10 -C.5296E-15	0.1184E 06 0.1184E C6	0.5225E 01 0.3170E 01	B2 33
NO	0.4147E 01 0.3221E C1 0.3343E C1	-0.4126E-02 0.9692E-05 0.1221E-02 -0.4297E-C6	-0.7863E-08 0.2231E-11 0.6559E-10 -0.3451E-14	0.9745E 04 0.9764E 04	0.2569E 01 0.6510E 01	*M1 B2
NO2	0.3434E 01 0.4614E C1	0.2223E-02 0.6715E-05 0.2636E-02 -0.1995E-05	-0.9743E-08 0.3721E-11 0.2082E-09 -0.1465E-13	0.2865E C4 0.2340E 04	0.8408E 01 0.1368E 01	*M1 *M2
NS	0.4262E C1 0.3840E C1	-0.2819E-02 0.9316E-C5 0.7457E-03 -0.3058E-C6	-0.9509E-08 0.2284E-11 0.5805E-10 -0.4069E-14	0.3087E 05 0.3079E 05	0.4061E 01 0.4472E 01	*M1 *M2
N2+	0.3397E 01 0.3370E C1	0.4525E-C3 0.1272E-06 0.8629E-03 -C.1276E-06	-0.3879E-10 0.2459E-14 0.8037E-11 -0.1330E-15	0.1826E 06 0.1826E 06	0.4205E 01 0.4073E 01	B2 33
N2	0.3692E C1 0.3221F 01 0.3727E 01	-0.1333E-02 0.2650E-05 0.9278E-03 -C.2907E-06	-0.9769E-09 -0.9977E-13 0.3938E-10 -C.2030E-14	-0.1063E 04 -0.1043E 04	0.2287E 01 0.4326E 01	*M1 B2
N20	0.2332E C1 0.4627F 01	0.1035F-01 -0.1117E-04 0.3922E-02 -0.1216E-05	0.6958E-08 -0.1878E-11 0.2286E-09 -C.1585E-13	0.8723E 04 0.8136E C4	0.1023E 02 -0.1146E 01	*N1 *M2
N204	0.3165E C1 0.1043E C1	0.2719E-01 -C.2535E-04 0.6036E-02 -C.2583E-C5	0.1099E-07 -0.1660E-11 0.4928E-09 -C.3478E-13	-0.7673E C3 -0.2741E C4	0.11148E 02 -0.2582E 02	*M1 *M2

N E +	C . 2705E 01	-0 . 9724E -04	0 . 1205E -07	0 . 5333E -12	-0 . 1213E -15	0 . 2252E 06	0 . 3716E C1	B2
	0 . 2664E 01	-0 . 6497E -04	0 . 9630E -08	-0 . 6142E -12	0 . 1413E -16	0 . 2252E 06	0 . 3961E 01	33
N E ++	C . 2318E 01	-0 . 2139E -03	0 . 5385E -07	-0 . 5739E -11	0 . 2467E -15	0 . 7276E 06	0 . 3424E 01	B2
	0 . 2787E 01	-0 . 1306E -03	0 . 3031E -07	-0 . 1952E -11	0 . 4256E -15	0 . 7276E 06	0 . 3588E 01	33
N E +++	C . 2501E 01	-0 . 2536E -05	C . 2462E -08	-0 . 8359E -12	0 . 9277E -16	0 . 1468E C7	0 . 4726E 01	B2
	0 . 2505E 01	0 . 1709E -04	-C . 1147E -07	0 . 1808E -11	-0 . 5856E -16	0 . 1468E 07	0 . 4681E 01	33
N E	C . 2500E 01	0 . 9636E -06	-C . 5972E -09	0 . 1288E -12	-0 . 8901E -17	-0 . 7454E 03	0 . 3684E 01	B2
	0 . 2571E 01	-0 . 6431E -04	0 . 1585E -07	-0 . 1469E -11	0 . 4619E -16	-0 . 7454E 03	0 . 2918E 01	33
O +	C . 2491E 01	0 . 2762E -04	-C . 1881E -07	0 . 3807E -11	-0 . 1028E -15	0 . 1879E 06	0 . 4424E 01	32
	0 . 2944E 01	-0 . 4108E -03	C . 9156E -07	-0 . 5848E -11	0 . 1190E -15	0 . 1879E 06	0 . 1750E 01	33
O + +	C . 2983E 01	-0 . 5800E -03	0 . 2334E -06	-0 . 3768E -10	0 . 2168E -14	0 . 5961E 06	0 . 2307E 01	B2
	0 . 2706E 01	-0 . 1315E -03	0 . 3049E -07	-0 . 2267E -11	0 . 5305E -16	0 . 5961E 06	0 . 3812E 01	33
O + + *	C . 3015E 01	-0 . 5885E -03	0 . 2355E -06	-0 . 3875E -10	0 . 2226E -14	0 . 1234E 07	0 . 1668E 01	B2
	0 . 2683E 01	-0 . 7049E -04	C . 9853E -08	-0 . 6203E -12	0 . 1678E -16	0 . 1234E 07	C . 3482E 01	53
O	C . 3022E 01	-0 . 2174E -02	0 . 3754E -05	-0 . 2995E -08	0 . 9078E -12	0 . 2914E 05	0 . 2646E 01	*M1
	0 . 2670E 01	-0 . 197CE -03	0 . 7193E -07	-0 . 8901E -11	0 . 4002E -15	0 . 2915E C5	C . 4504E 01	B2
C . 2545E 01	-0 . 5952E -04	0 . 2701E -07	-0 . 2798E -11	0 . 9386E -16	0 . 2915E 05	C . 5049E 01	33	
O -	C . 2500E 01	0 . 344CE -06	-0 . 1954E -09	0 . 3937E -13	-0 . 2573E -17	0 . 1141E 05	0 . 4786E 01	B2
	0 . 2508E 01	-0 . 6332E -05	0 . 1364E -08	-0 . 1094E -12	0 . 2934E -17	0 . 1141E 05	0 . 4739E 01	33
O H	C . 3323E 01	-C . 1119E -02	0 . 1247E -05	-0 . 2104E -09	-C . 5255E -13	0 . 3585E C4	0 . 5825E 00	*M1
	C . 3284E 01	C . 5037E -03	-C . 2289E -07	-0 . 1001E -10	0 . 1056E -14	0 . 3626E C4	C . 3232E 01	B2
	C . 3456E 01	C . 4375E -03	-0 . 5819E -07	0 . 4187E -11	-0 . 1104E -15	0 . 3626E C4	0 . 2149E C1	33

02+	0.3243E 0.5169E	01 01	0.1174E-02 -0.8620E-03	-C.3900E-06 0.2041E-C6	0.5437E-10 -0.1300E-10	-0.2392E-14 0.2494E-15	0.1400E 0.140CE	06 06	0.5925E -0.5296E	01 01	32 B3
02	0.3719E 0.3316E 0.3721E	01 01 01	-0.2517E-02 0.1151E-02 0.4254E-03	0.8584E-C5 -0.3726E-C6 -0.2835E-C7	-0.8300E-08 0.6186E-10 0.6050E-12	0.2708E-11 -0.3666E-14 -0.5186E-17	-0.1056E -C.1C44E -0.1C44H	C4 C4 C4	0.3903E 0.5393E C.3254E	01 01 01	*M1 B2 B3
02-	0.3402E 0.3648E	01 01	0.1243E-02 0.2944E-03	-0.5081E-C6 0.536CE-C7	0.9982E-10 -0.8739E-11	-0.6398E-14 0.2328E-15	-0.1063E -0.1063E	05 05	0.5428E 0.4575E	01 01	B2 B3

APPENDIX B

DETERMINATION OF THE HEATS OF FORMATION OF PURE
COMPOUNDS AT 298°K

As indicated in the introduction of this report, several sources used for the thermodynamic functions gave tabulated properties of enthalpy and free energy relative to elements at 0°K. In order to transform this reference temperature to 298.16, it is necessary to know the heat of reaction at the latter temperature. The method used can be best illustrated by Figure B-1.

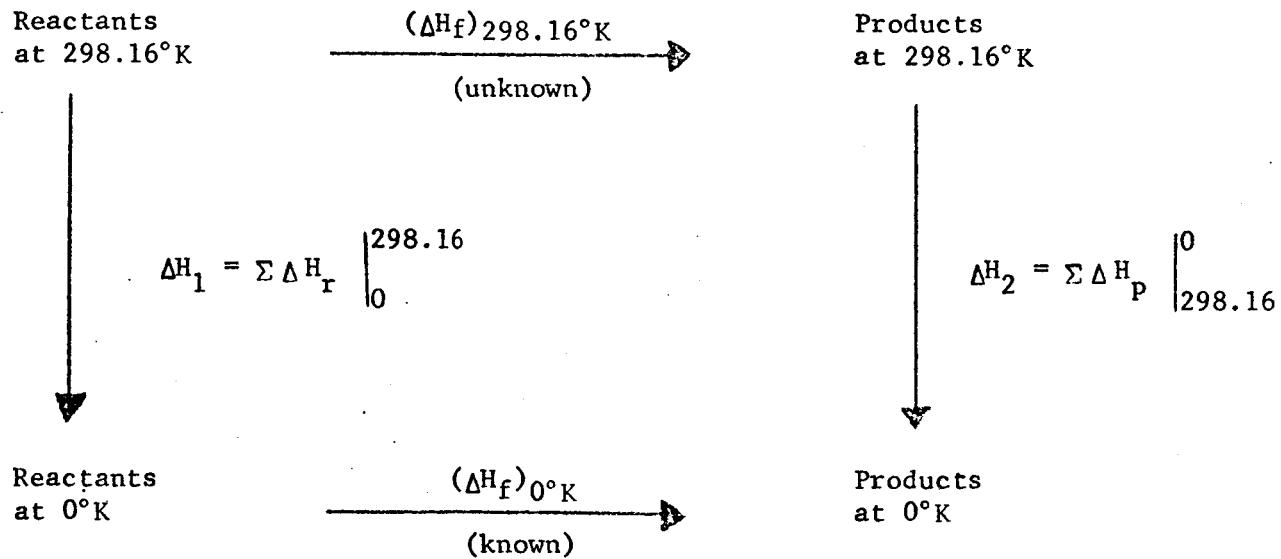


Figure B-1. Determination of Heat of Formation at 298.16°K from Heat of Formation at 0°K

By definition, the heat of formation of a compound is the enthalpy change incurred by its production. Since enthalpy is a point function and as such is independent of path, we can cool the reactants from the reference temperature of 0°K, allow the formation reaction to occur at the lower temperature, and then heat the products back to the new reference temperature of 298.16°K. Through this process we can establish the enthalpy of

the reaction products relative to the reactants or the heat of reaction at 298.16° by the following relation,

$$(\Delta H_f^\circ)_{298.16} = (\Delta H_f^\circ)_0 + \Delta H_1 + \Delta H_2 \quad (B-1)$$

Some care must be exercised in selecting the reaction for the formation of a particular compound. The enthalpy of the reactants must be specified relative to the reference species. It is therefore convenient to use the reference species as reactants in formation reaction. The reference elements used in this analysis are; H₂, N₂, O₂, Ne, Ar, He, carbon solid and e⁻. The reaction used for the formation of each compound are listed in Table B-1. Also included in this table are the heats of formation at 0°K and 298.16°K.

Table B-1. Reactions and Heats of Formation

Component	Reaction	$(\Delta H_f^\circ)_{298.16}$ (Kcal/gmole)	$(\Delta H_f^\circ)_0$ (Kcal/gmole)	Reference
N	$1/2 N_2 \rightarrow N$	112.951	112.507	2
O	$1/2 O_2 \rightarrow O$	59.544	58.972	2
C(gas)	$C \rightarrow C$	171.301	169.990	2
A ⁺	$A \rightarrow A^+ + e^-$	364.828	363.345	2
N ⁺	$1/2 N_2 \rightarrow N^+ + e^-$	449.709	447.564	2
O ⁺	$1/2 O_2 \rightarrow O^+ + e^-$	374.867	372.942	2
C ⁺	$C_s \rightarrow C^+ + e^-$	432.357	429.537	2
Ne ⁺	$Ne \rightarrow Ne^+ + e^-$	449.071	497.186	2
A ⁺⁺	$A \rightarrow A^{++} + 2e^-$	1003.252	1000.280	2
N ⁺⁺	$1/2 N_2 \rightarrow N^{++} + 2e^-$	907.179	905.023	2
O ⁺⁺	$1/2 O_2 \rightarrow O^{++} + 2e^-$	1186.408	1182.600	2
C ⁺⁺	$C_s \rightarrow C^{++} + 2e^-$	995.898	991.689	2
Ne ⁺⁺	$Ne \rightarrow Ne^{++} + 2e^-$	1447.366	1444.350	2
A ⁺⁺⁺	$A \rightarrow A^{+++} + 3e^-$	1947.934	1943.490	2
N ⁺⁺⁺	$1/2 N_2 \rightarrow N^{+++} + 3e^-$	2003.648	1998.760	2
O ⁺⁺⁺	$1/2 O_2 \rightarrow O^{+++} + 3e^-$	2454.520	2449.470	2
C ⁺⁺⁺	$C_s \rightarrow C^{+++} + 3e^-$	2101.403	2095.690	2
Ne ⁺⁺⁺	$Ne \rightarrow Ne^{+++} + 3e^-$	2918.493	2914.050	2
O ⁻	$1/2 O_2 + e^- \rightarrow O^-$	24.156	25.193	2
H	$1/2 H_2 \rightarrow H$	52.098	51.620	5
OH	$1/2 O_2 + 1/2 H_2 \rightarrow OH$	9.3575	9.300	5
H ₂ O	$1/2 O_2 + H_2 \rightarrow H_2O$	-57.8018	-57.107	5
HO ₂	$O_2 + 1/2 H_2 \rightarrow HO_2$	5.0244	5.700	5
HCO	$C_2 + 1/2 O_2 + 1/2 H_2 \rightarrow HCO$	-3.224	-3.311	9
H ₂ CO	$C_s + 1/2 O_2 + H_2 \rightarrow H_2 CO$	-27.8168	-26.900	5
HCO ⁺	$C_s + 1/2 O_2 + 1/2 H_2 \rightarrow HCO^+ + e^-$	206.5688	205.000	5
H ₃ O ⁺	$O_2 + 3H_2 \rightarrow 2H_3O^+ + 2e^-$	140.6009	141.000	5
C ₂	$C_s \rightarrow C_2$	197.0259	195.000	5
CH	$C_s + 1/2 H_2 \rightarrow CH$	142.3965	141.600	5
CH ₂	$C_s + H_2 \rightarrow CH_2$	94.8097	95.000	5
CH ₃	$2C_s + 3H_2 \rightarrow 2CH_3$	31.8032	34.400	5
CH ₄	$C_s + 2H_2 \rightarrow CH_4$	-17.8964	-15.990	5

Table B-1. (cont.)

Component	Reaction	$(\Delta H_f^\circ)_{298.16}$ (Kcal/gmole)	$(\Delta H_f^\circ)_0$ (Kcal/gmole)	Reference
C ₂ H	2C _s + 1/2 H ₂ → C ₂ H	117.6488	116.700	5
C ₂ H ₂	2C _s + H ₂ → C ₂ H ₂	53.8670	54.330	5
C ₂ H ₃	4C _s + 3H ₂ → 2C ₂ H ₃	64.9850	66.900	5
C ₂ H ₄	2C _s + 2H ₂ → C ₂ H ₄	14.4926	14.520	5
C ₄ H ₂	4C _s + H ₂ → C ₄ H ₂	111.7205	111.300	5
NO	1/2 N ₂ + 1/2 O ₂ → NO	21.6009	21.477	3
CO	C _s + 1/2 O ₂ → CO	-26.4179	-27.202	3
O ₂ ⁺	O ₂ → O ₂ ⁺ + e ⁻	280.2099	277.918	3
N ₂ ⁺	N ₂ → N ₂ ⁺ + e ⁻	364.9392	359.306	3
NO ⁺	1/2 N ₂ + 1/2 O ₂ → NO ⁺ + e ⁻	237.3239	235.836	3
CO ⁺	C _s + 1/2 O ₂ → CO ⁺ + e ⁻	298.2493	295.977	3
O ₂ ⁻	e ⁻ + O ₂ → O ₂ ⁻	-19.0502	-23.000	3
CN	C _s + 1/2 N ₂ → CN	109.7865	109.000	4
C ₂ N ₂	2C _s + N ₂ → C ₂ N ₂	74.1277	73.400	4
C ₃	3C _s → C ₃	189.6115	188.000	4
C ⁻	C _s + e ⁻ → C ⁻	142.2300	141.000	4

APPENDIX C
LISTING OF COMPUTER PROGRAM

RTRAN IV G LEVEL 18

MAIN

DATE = 71062

18/05/46

C *
C PROGRAM FOR CURVE-FIT OF THERMODYNAMIC FUNCTIONS
C

C THIS PROGRAM CALCULATES THE VALUES OF THE CONSTANTS A1-A7 FOR
C THE POLYNOMIAL FORMS OF THE FREE ENERGY, ENTHALPY, HEAT
C CAPACITY, AND ENTROPY RELATIONSHIPS AS USED IN MCBRIDE, ET AL.,
C NASA SP-3001, PP. 14-15.
C

C THE FOLLOWING INPUT DATA ARE REQUIRED*

1. IDENTIFICATION CARD-- CONSISTS OF SPECIES NAME IN COLS. 1-8, REFERENCE TEMPERATURE IN COLS. 16-22, DELTA H OF FORMATION AT THE REFERENCE TEMPERATURE (CAL/MOLE) IN COLS. 32-46, AND THE VALUE OF THE ENTHALPY FUNCTION AT THE REFERENCE TEMPERATURE (CAL/MOLE) IN COLS. 51-60. ALL NUMERICAL ENTRIES ON THIS CARD ARE TO BE IN F-FORMAT.
2. SOURCE CARD -- COLS. 1-79 ARE AVAILABLE FOR A BRIEF BIBLIOGRAPHICAL CITATION.
3. FUNCTION DATA CARDS -- THESE CARDS CONTAIN TEMPERATURES IN COLS. 1-6, AND THE CORRESPONDING VALUES OF THE FREE ENERGY AND ENTHALPY FUNCTIONS RESPECTIVELY IN COLS 7-20 AND 21-34. THE FORM OF THESE FUNCTIONS AS REQUIRED FOR INPUT CAN BE VARIED. IN GENERAL, TABULATED DATA IS OF THE FORM OF $-(F-H^*)/T$. IF THE DATA IS OF THE FORM OF $-(F-H^*)$, THE NUMBER 1 CAN BE INSERTED IN COL. 80 OF THE SOURCE CARD AND THE INPUT WILL BE TRANSFORMED INTERNALLY TO THE GENERAL FORM.
4. TERMINATION CARD--A NEGATIVE VALUE IN COLS. 1-6 WILL TERMINATE DATA INPUT. AFTER THIS PACKAGE HAS BEEN PROCESSED A NEW SET OF CARDS WILL BE READ.

DONALD D. ESCH
LOUISIANA STATE UNIVERSITY
AUGUST 1970

C *
001 DOUBLE PRECISION T(100),H(100),B(7)
002 DIMENSION SOURCE(39),SP(2),F(100),HC(100),FC(100),A(7)
003 DIMENSION S(100),CP(100),HH(100)
004 INTEGER RANGE
005 R=1.98718
006 10 I=1
007 READ 100,SP,TRER,DELH,HREF
008 READ 101,SOURCE,NTYPE
009 20 READ 102,T(I),F(I),H(I)
010 IF(T(I).LT.0.0)GOTO30
011 I=I+1
012 GOTO20
013 30 NUMBER=I-1
014 TREF=298.16
C
C-----SET UP ARRAYS FOR LEAST SQUARES FIT.
C
015 DO50 I=1,NUMBER

RTRAN IV G LEVEL 18

MAIN

DATE = 71062

18/05/4

```

016      H(I)=H(I)/R
017      50   F(I)=-F(I)/R
018      IF(NTYPE.EQ.0)GOTO70
019      DO60I=1,NUMBER
020      H(I)=H(I)/T(I)
021      60   F(I)=F(I)/T(I)
022      70   CONTINUE
C
023      CALL LSFIT(T,H,NUMBER,4,B)
C
024      A(1)=B(1)
025      A(2)=B(2)*2.
026      A(3)=B(3)*3.
027      A(4)=B(4)*4.
028      A(5)=B(5)*5.
C
C-----DETERMINE THE VALUE OF A(6)
C
029      HZERO=DELH-HREF
030      A(6)=HZERO/R
C
C-----DETERMINE THE VALUE OF A(7)
C
031      SUMA7=0.0
032      DO80I=1,NUMBER
033      TEMP=T(I)
034      FEST=A(1)*(1.- ALOG(TEMP))-A(2)*T(I)/2.-A(3)*(T(I)**2)/6.-A(4)*
1 (T(I)**3)/12.-A(5)*(T(I)**4)/20.
035      DF=FEST-F(I)
036      80   SUMA7=SUMA7+DF
037      A(7)=SUMA7/NUMBER
C
C-----DETERMINE TEMPERATURE RANGE---RANGE 1 INCLUDE TEMPERATURES FROM
C      0 TO 1000, RANGE 2 FROM 1000 TO 6000, AND RANGE 3 FROM 6000 TO
C      18000 DEGREES KELVIN.
C
038      RANGE=1
039      IF(T(NUMBER).LE.1000)GOTO85
040      RANGE=2
041      IF(T(NUMBER).LE.8000)GOTO85
042      RANGE=3
043      85   CONTINUE
C
C-----OUTPUT
C
C-----PUNCH OUT THERMO DATA CARDS FOR CHEMEQ PROGRAM
C
C      DO88N=1,20
088      WRITE(7,117)(A(I),I=1,7),SP,RANGE
C
044      PRINT 113
045      PRINT 103,SP
046      IF (RANGE.LT.3)GOTO47
047      IF (T(NUMBER).EQ.1000) GO TO 51
048      ITLO=6000
049      ITHI=15000
050      JANE=6
051      JIM=24

```

TRAN IV G LEVEL 18

MAIN

DATE = 71062

18/05/46

```

052      GO TO 49
053      51 ITLO=6000
054      ITHI=10000
055      JANE=6
056      JIM=26
057
058      GOT049

059      47 ITLO=1000
060      ITHI=6000
061      JANE=1
062      JIM=26
063      49 PRINT 104,ITLO,ITHI
064      PRINT 105,TREF
065      PRINT 106
066      PRINT 108
067      D090I=JANE,JIM
068      TEMP=T(I)
069      FC(I)=A(1)*(1.-ALOG(TEMP))-A(2)*T(I)/2.-A(3)*(T(I)**2)/6.
070      1 - A(4)*(T(I)**3)/12. - A(5)*(T(I)**4)/20. - A(7)
071      FC(I)=-FC(I)
072      HC(I)=A(1) + A(2)*T(I)/2. + A(3)*(T(I)**2)/3. + A(4)*(T(I)**3)/4.
073      1 + A(5)*(T(I)**4)/5.
074      CP(I)=A(1) + A(2)*T(I) + A(3)*(T(I)**2) + A(4)*(T(I)**3) +
075      & A(5)*(T(I)**4)
076      S(I)=A(1)*ALOG(TEMP) + A(2)*T(I) + A(3)*(T(I)**2)/2.
077      & + A(4)*(T(I)**3)/3. + A(5)*(T(I)**4)/4. + A(7)
078      HTO=HC(I) + (A(6)/T(I))
079      NT=T(I)
080      90 PRINT 112,NT,FC(I),HC(I),S(I),CP(I),HTO
081      PRINT 108
082      PRINT 109
083      PRINT 110,(A(I),I=1,7)
084      PRINT 111,SOURCE
085      NN=JIM-JANE
086      NCOUNT=26-NN
087      IF(NCOUNT.LT.1)NCOUNT=1
088      D095N=1,NCOUNT
089
090      95 PRINT 116
091      PRINT 114
092      WRITE(6,117)(A(I),I=1,7),SP,RANGE
093      PRINT 115
094      GOT010
095      200 STOP
C
C----- F O R M A T   S T A T E M E N T S
C
096      100 FORMAT(2A3,9X,F6.2,10X,F14.5,5X,F10.2)
097      101 FORMAT(39A2,I2)
098      102 FORMAT(F6.0,2E14.4)
099      103 FORMAT(25X,'POLYNOMIAL CONSTANTS AND THERMODYNAMIC PROPERTIES OF '
1 ,2A3)
100      104 FORMAT(25X,'OVER A TEMPERATURE RANGE OF ',I5,' TO ',I5,' DEGREES K
1ELVIN.')
101      105 FORMAT(37X,'REFERENCE TEMPERATURE = ',F6.2,/)
102      106 FORMAT(14X,'T(K)',5X,'-(F-H*)/RT',7X,'(H-H*)/RT',10X,'S/R',13X,'C
1P/R',11X,'H/RT')

```

RTRAN IV G LEVEL 18

MAIN

DATE = 71062

18/05/45

```
098 108 FORMAT(10X,18('-----'),/)  
099 109 FORMAT(  
110     16X,'A1',11X,'A2',11X,'A3',11X,'A4',11X,'A5',  
111     1 11X,'A6',11X,'A7')  
100 110 FORMAT(10X,7(E13.6),/)  
101 111 FORMAT(8X,' SOURCE OF DATA* ',5X,3A2)  
102 112 FORMAT(10X,I8,5E16.6)  
103 113 FORMAT(1H1,/,22('-----'),7(/,' ',108X,' '))  
104 114 FORMAT(  
115     22('-----'))  
105 115 FORMAT(1H1)  
106 116 FORMAT(' ',108X,' ')  
107 117 FORMAT(7E10.4,2X,2A3,1X,I1)  
108 120 FORMAT(F20.2)  
109      END
```

IRTRAN IV G LEVEL 18

LSFIT

DATE = 71062

18/05/46

```

001      SUBROUTINE LSFIT(X,Y,NUMBER,M,C)
C
C      LEAST SQUARE CURVE FITTING OF ANY ORDER POLYNOMIAL
C      OF ORDER EQUAL TO OR LESS THAN 10
C
C
C      NUMBER IS THE ACTUAL NUMBER OF X-Y DATA PAIRS.,MAXIMUM OF 200
C      M IS THE DEGREE OF THE POLYNOMIAL.,MAXIMUM OF 10
C      N IS THE NUMBER OF EQUATIONS(=M+1)
C      X,Y IS THE ARRAY FOR THE DATA PAIRS
C      A IS THE ARRAY FOR THE SUM, WHICH BECOME THE COEFFICIENTS OF THE
C      UNKNOWN IN THE SIMULTANEOUS EQUATIONS.
C      B IS THE ARRAY FOR THE CONSTANT TERMS IN THE SIMULTANEOUS EQUATIONS
C      C IS THE ARRAY FOR THE UNKNOWNS, WHICH BECOME THE COEFFICIENTS IN
C      THE POLYNOMIAL.
C      P IS THE ARRAY FOR THE POWERS OF THE X(I), FROM 1 TO 2M.
C
002      REAL*8           X(100),Y(100),A(7,7),B(7),C(7),P(20)
C
003      MX2=M*2
004      DO 13 I=1,MX2
005      P(I)=0.0
006      DO 13 J=1,NUMBER
007      POWER=I
008      13 P(I)=P(I)+X(J)**POWER
C
C      DEVELOPING THE COEFFICIENTS AND THE CONSTANT TERMS OF THE NORMAL
C      EQUATIONS.
C
009      N=M+1
010      DO 30 I=1,N
011      DO 30 J=1,N
012      K=I+J-2
013      IF(K)29,29,28
014      28 A(I,J)=P(K)
015      GO TO 30
016      29 A(1,1)=NUMBER
017      30 CONTINUE
018      B(1)=0.
019      DO 21 J=1,NUMBER
020      21 B(1)=B(1)+Y(J)
021      DO 22 I=2,N
022      B(I)=0.
023      DO 22 J=1,NUMBER
024      22 B(I)=B(I)+Y(J)*X(J)**(I-1)
C
C      PIVOTAL CONDENSATION
C
025      NM1=N-1
026      DO 300 K=1,NM1
027      KP1=K+1
028      L=K
029      DO 400 I=KP1,N
030      IF(DABS(A(I,K))-DABS(A(L,K)))400,400,401

```

TRAN IV G LEVEL 18

LSFIT

DATE = 71062

18/05/46

```

31      401 L=I
32      400 CONTINUE
33      IF(L-K)500,500,405
34      405 DO 410 J=K,N
35          TEMP=A(K,J)
36          A(K,J)=A(L,J)
37      410 A(L,J)=TEMP
38          TEMP=B(K)
39          B(K)=B(L)
40          B(L)=TEMP

```

C
 C ELIMINATION, BACK SOLUTION, AND PRINTING RESULTS
 C

```

41      500 DO 300 I=KP1,N
42          FACTOR=A(I,K)/A(K,K)
43          A(I,K)=0.0
44          DO 301 J=KP1,N
45          301 A(I,J)=A(I,J)-FACTOR*A(K,J)
46          300 B(I)=B(I)-FACTOR*B(K)
47          C(N)=B(N)/A(N,N)
48          I=N-1
49      710 IP1=I+1
50          SUM=0.0
51          DO 700 J= IP1,N
52          700 SUM=SUM+A(I,J)*C(J)
53          C(I)=(B(I)-SUM)/A(I,I)
54          I=I-1
55          IF(I)800,800,710
56      800 RETURN
57      END

```

APPENDIX D
TABULATED THERMODYNAMIC PROPERTIES

In this appendix values of free energy, enthalpy, heat capacity and entropy, as predicted by the polynomial coefficients listed in Appendix A, are given in tabular form. Each table contains, in addition to the above tabulations, the corresponding polynomial constants and the original source of the data from which the constants were evaluated.

POLYNOMIAL CONSTANTS AND THERMODYNAMIC PROPERTIES OF A+
 OVER A TEMPERATURE RANGE OF 1000 TO 6000 DEGREES KELVIN.
 REFERENCE TEMPERATURE = 298.16

T(K)	$-(F-H^*)/RT$	$(H-H^*)/RT$	S/R	CPR	H/RT
1000	0.205521E 02	0.263107E 01	0.231832E 02	0.268569E 01	0.185476E 03
1200	0.210327E 02	0.264051E 01	0.236732E 02	0.268865E 01	0.155011E 03
1400	0.214403E 02	0.264723E 01	0.240875E 02	0.268557E 01	0.133251E 03
1600	0.217941E 02	0.265158E 01	0.244457E 02	0.267774E 01	0.116930E 03
1800	0.221065E 02	0.265388E 01	0.247604E 02	0.266631E 01	0.104234E 03
2000	0.223862E 02	0.265444E 01	0.250406E 02	0.265233E 01	0.940768E 02
2200	0.226392E 02	0.265355E 01	0.252927E 02	0.263677E 01	0.857648E 02
2400	0.228700E 02	0.265148E 01	0.255214E 02	0.262049E 01	0.788368E 02
2600	0.230821E 02	0.264847E 01	0.257306E 02	0.260424E 01	0.729734E 02
2800	0.232782E 02	0.264474E 01	0.259230E 02	0.258867E 01	0.679465E 02
3000	0.234606E 02	0.264052E 01	0.261011E 02	0.257436E 01	0.635888E 02
3200	0.236308E 02	0.263598E 01	0.262668E 02	0.256175E 01	0.597750E 02
3400	0.237905E 02	0.263129E 01	0.264219E 02	0.255120E 01	0.564092E 02
3600	0.239408E 02	0.262660E 01	0.265674E 02	0.254296E 01	0.534168E 02
3800	0.240827E 02	0.262204E 01	0.267047E 02	0.253719E 01	0.507391E 02
4000	0.242170E 02	0.261770E 01	0.268347E 02	0.253394E 01	0.483289F 02
4200	0.253581E 02	0.259561E 01	0.279537E 02	0.257947E 01	0.320867E 02
4400	0.254405E 02	0.259503E 01	0.280355E 02	0.257411E 01	0.311645E 02
4600	0.245821E 02	0.260685E 01	0.271889E 02	0.253836E 01	0.423557E 02
4800	0.246930E 02	0.26041CE 01	0.272971E 02	0.254373E 01	0.406968E 02
5000	0.247992E 02	0.260182E 01	0.274011E 02	0.255038E 01	0.391708E 02
5200	0.249013E 02	0.259998E 01	0.275012E 02	0.255777E 01	0.377624E 02
5400	0.249993E 02	0.259856E 01	0.275979E 02	0.256525E 01	0.364587E 02
5600	0.250938E 02	0.259749E 01	0.276913E 02	0.257205E 01	0.352483E 02
5800	0.251850E 02	0.259671E 01	0.277817E 02	0.257735E 01	0.341217F 02
6000	0.252730E 02	0.259612E 01	0.278691E 02	0.258017E 01	0.330703E 02

A1 A2 A3 A4 A5 A6 A7
 0.252945E 01 0.315090E-03-0.196105E-06 0.398768E-10-0.261838E-14 0.182845E 06 0.548071E 01

SOURCE OF DATA* W.G. BROWNE (ENGR PHYS TECH MEMO-2) APPENDIX A

REMAINDER OF APPENDIX D AVAILABLE ON REQUEST