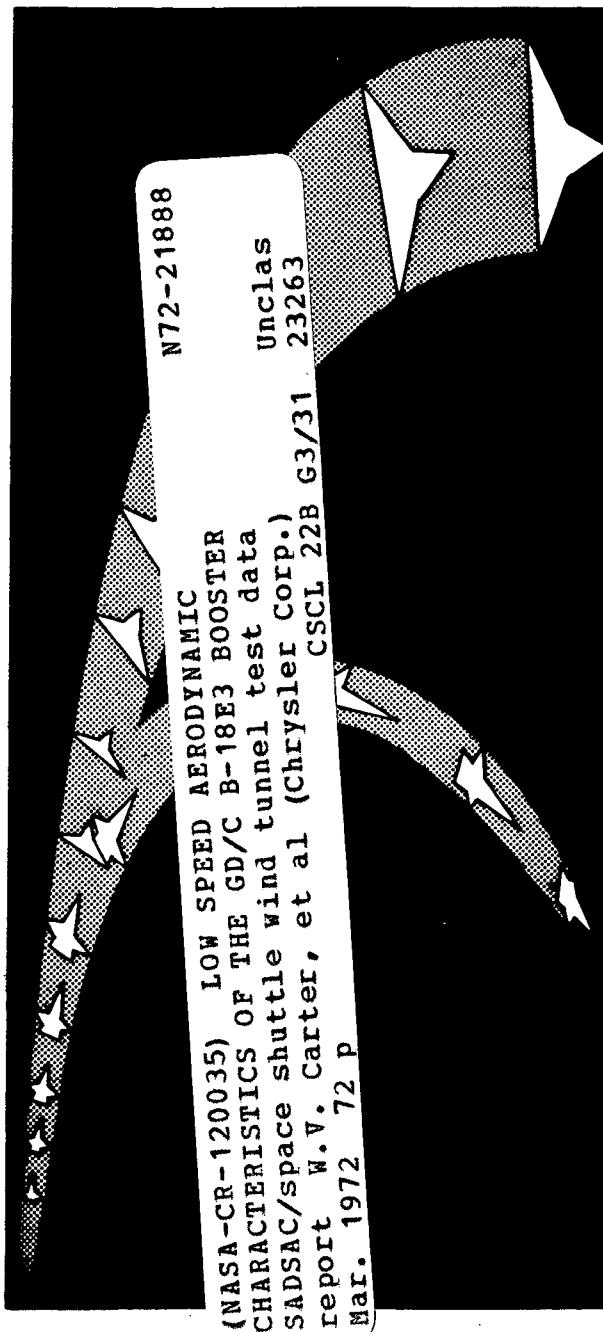


23-17  
DMS-DR-1223  
CR-120,035  
MARCH 1972

-SPACE SHUTTLE-



SADSAC SPACE SHUTTLE  
AEROTHERMODYNAMIC  
DATA MANAGEMENT SYSTEM

CONTRACT NAS8-4016

MARSHALL SPACE FLIGHT CENTER

SPACE DIVISION



This document should be  
referenced as NASA CR-120,035

CAT. 31

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DMS-DR-1223  
CR-120,035  
March, 1972

SADSAC/SPACE SHUTTLE

WIND TUNNEL TEST DATA REPORT

CONFIGURATION: GD/Convair 0.02 Scale B-18E3 Booster

TEST PURPOSE: Investigate Low Speed Longitudinal and Lateral Directional  
Characteristics

TEST FACILITY: GD Low Speed Wind Tunnel

TESTING AGENCY: GD/SDO

TEST NO. & DATE: GDLST 603-0; December 15-17, 1971

FACILITY COORDINATOR: R. Lanflesi - GD/C

PROJECT ENGINEER(S): W. V. Carter - GD/C

W. H. Gallagher - GD/C

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CONTRACT NAS 8-4016

AMENDMENT 153

DRL 184 - 58

This report has been prepared by Chrysler Corporation Space Division under a Data Management Contract to the NASA. Chrysler assumes no responsibility for the data presented herein other than its display characteristics.

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LOW SPEED AERODYNAMIC CHARACTERISTICS OF  
THE GD/C B-18E3 BOOSTER

By W. V. Carter and W. H. Gallaher

ABSTRACT

A 0.02 scale model of the General Dynamics Corporation B-18E3 Space Shuttle Booster was tested in the GD Low Speed Wind Tunnel to evaluate the low speed aerodynamic characteristics. The basic configuration, including build-up, was tested at a Mach number of 0.201 and Reynolds number per foot of 1.39 million. The normal angle-of-attack range was -4 to +24 degrees in 2 degree increments, at sideslip angles of 0 and 5 degrees. Some lateral data was obtained at the sideslip angle range of -6 to 10 degrees at angles-of attack of 0, 10 and 15 degrees. Data was obtained for canard, split elevon, and split rudder deflections.

## NOMENCLATURE

### (General)

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
$\alpha$	ALPHA	angle of attack, angle between the projection of the wind $X_w$ -axis on the body X, Z-plane and the body X-axis; degrees
$\beta$	BETA	sideslip angle, angle between the wind $X_w$ -axis and the projection of this axis on the body X-Z-plane; degrees
$\psi$	PSI	yaw angle, angle of rotation about the body Z-axis, positive when the positive X-axis is rotated toward the positive Y-axis; degrees
$\phi$	PHI	roll angle, angle of rotation about the body X-axis, positive when the positive Y-axis is rotated toward the positive Z-axis; degrees
$\rho$		air density; $\text{kg/m}^3$ , slugs/ $\text{ft}^3$
$a$		speed of sound; $\text{m/sec}$ , $\text{ft/sec}$
$V$		speed of vehicle relative to surrounding atmosphere; $\text{m/sec}$ , $\text{ft/sec}$
$q$	$Q(\text{PSI})$ $Q(\text{PSF})$	dynamic pressure; $1/2\rho V^2$ , psi, psf
$M$	MACH	Mach number; $V/a$
$RN/L$	RN/L	Reynolds number per unit length; million/ $\text{ft}$
$p$		static pressure; psi
$P$		total pressure; psi
$C_p$	CP	pressure coefficient; $(p-p_\infty)/q$

## NOMENCLATURE (Continued)

### Reference & C. G. Definitions

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
S		wing area; $\text{m}^2$ , $\text{ft}^2$
S	SREF	reference area; $\text{m}^2$ , $\text{ft}^2$
$\bar{c}$		wing mean aerodynamic chord or reference chord; m, ft, in (see $l_{ref}$ or LREF)
$l_{ref}$	LREF	reference length; m, ft, in.; (see $\bar{c}$ )
$b_{ref}$	BREF	wing span or reference span; m, ft, in
$A_b$		base area; $\text{m}^2$ , $\text{ft}^2$ , $\text{in}^2$
c. g.		center of gravity
MRP	MRP	abbreviation for moment reference point
	XMRP	abbreviation for moment reference point on X-axis
	YMRP	abbreviation for moment reference point on Y-axis
	ZMRP	abbreviation for moment reference point on Z-axis

## NOMENCLATURE (Continued)

### Axis System General

<u>SYMBOL</u>	<u>DEFINITION</u>
F	force; F, lbs
M	moment; M, in-lb

<u>Subscript</u>	<u>Definition</u>
N	normal force
A	axial force
L	lift force
D	drag force
Y	force or moment about the Y axis
Z	moment about the Z axis
X	moment about the X axis
S	stability axis system
W	wind axis system
ref	reference conditions
$\infty$	free stream conditions
t	total conditions
b	base

NOMENCLATURE (Continued)  
Body & Stability Axis System

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
<u>Body Axis System</u>		
$C_N$	CN	normal force coefficient; $F_N/qS$
$C_A$	CA	axial force coefficient; $F_A/qS$
$C_{A_b}$	CAB	base axial force coefficient; $[-1] \left[ (p_b - p_\infty)/q \right] (A_b/S)$
$C_{A_f}$	CAF	forebody axial force coefficient; $C_A - C_{A_b}$
$C_n$	CYN	yawing moment coefficient; $M_Z/qS b_{ref}$
$C_l$	CBL	rolling moment coefficient; $M_X/qS b_{ref}$
<u>Common to Both Axis Systems</u>		
$C_m$	CLM	pitching moment coefficient; $M_Y/qS l_{ref}$
$C_y$	CY	side force coefficient; $F_Y/qS$
<u>Stability Axis System</u>		
$C_L$	CL	lift force coefficient; $F_L/qS$
$C_D$	CD	drag force coefficient; $F_D/qS$
$C_{D_b}$	CDB	base drag coefficient
$C_{D_f}$	CDF	forebody drag coefficient; $C_D - C_{D_b}$
$C_n$	CLN	yawing moment coefficient; $M_{Z,s}/qS b_{ref}$
$C_l$	CSL	rolling moment coefficient; $M_{X,s}/qS b_{ref}$
L/D	L/D	lift-to-drag ratio; $C_L/C_D$
L/D <sub>f</sub>	L/DF	lift to forebody drag ratio; $C_L/C_{D_f}$

ADDITIONS TO SADSAC NOMENCLATURE  
FOR  
GD LST TEST NO. 603-0

<u>SYMBOL</u>	SADSAC <u>SYMBOL</u>	<u>DEFINITION</u>
$\delta_{e_{XY}}$	DEXY	Deflection angle of split elevons in degrees, positive trailing edge down with notation:  X = L left side X = R right side Y = Ø outboard elevon Y = M middle elevon Y = I inboard elevon Y = blank full span elevon deflection  DEØ, DEM, DEI denote symmetrical deflections of outboard, middle, and inboard split elevons, respectively.
$\delta_e$	ELEVTR	Combined elevator deflection in degrees, for split elevons, $(DELØ + DELM + DELI + DERØ + DERM + DERI)/6$  or for full span deflections, $(DEL + DER)/2$
$\delta_a$	AILRØN	Combined aileron deflection in degrees, for split elevon, $[(DELØ + DELM + DELI) - (DERØ + DERM + DERI)]/6$  or for full span deflections, $(DEL-DER)/2$
$\delta_c$	CANARD	Canard deflection angle, positive trailing edge down.
$\delta_{RX}$	RUDDER	Rudder deflection angle, positive trailing edge left with notation:  X = U upper of split rudder X = L lower of split rudder X = blank, full span rudder deflection

## CONFIGURATIONS INVESTIGATED

### Configuration Nomenclature

The model components tested are described below:

<u>Symbol</u>	<u>Description</u>
B <sub>30</sub>	Basic B-18E3 booster configuration with R <sub>5</sub> rocket engines (4 engines with center engine removed) mounted at all times, canard fairings, canopy and X <sub>4</sub> rocket engine shrouds.
B <sub>31</sub>	Same as B <sub>30</sub> except with blunt nose
B <sub>32</sub>	8-Ft fuse. plug removed from B <sub>30</sub>
W <sub>23</sub>	Basic B-18E3 wing, 8549 ft <sup>2</sup> , AR = 2.436, $\lambda$ = .2158, 53° LE sweep, spanwise split elevons.
C <sub>10</sub>	Basic delta canard, 485 ft <sup>2</sup> full scale planform area
C <sub>11</sub>	Delta canard 364 ft <sup>2</sup> full scale planform area (75% of C <sub>10</sub> )
V <sub>14</sub>	Basic vertical tail, 1300 ft <sup>2</sup> full scale area, $\Lambda_{LE} = 40^\circ$ , AR = 1.4
V <sub>16</sub>	Vertical tail, 2200 ft <sup>2</sup> full scale area, $\Lambda_{LE} = 40^\circ$
E <sub>38</sub>	Air breathing engines arrangement. 10 flow through nacelles individually mounted beneath wing and body
X <sub>5</sub>	Enlarged sting support
X <sub>6</sub>	Indicates the gap between the inboard elevon and body was sealed
X <sub>7</sub>	Indicates that the entire L.E. region of wing W <sub>23</sub> was covered with 036 grit.

### Combinations Tested

The basic B-18E3 configuration consisted of the following components:

B<sub>30</sub> W<sub>23</sub> C<sub>10</sub> V<sub>14</sub> E<sub>38</sub>

with grit applied to the model nose, and near the leading edges of the canard, wing and vertical tail. The following combinations were tested:

<u>Combination</u>	<u>Run Numbers</u>
B <sub>30</sub>	93, 94
B <sub>32</sub>	95, 96
B <sub>30</sub> V <sub>14</sub>	91, 92
B <sub>30</sub> C <sub>10</sub>	83 to 90
B <sub>30</sub> W <sub>23</sub> E <sub>38</sub>	65, 66
B <sub>30</sub> W <sub>23</sub> V <sub>14</sub>	77, 78
B <sub>30</sub> W <sub>23</sub> V <sub>16</sub>	79, 80
B <sub>30</sub> W <sub>23</sub> V <sub>14</sub> E <sub>38</sub>	44, 45
B <sub>30</sub> W <sub>23</sub> C <sub>10</sub> V <sub>16</sub>	81, 82
B <sub>30</sub> W <sub>23</sub> C <sub>10</sub> V <sub>14</sub>	69, 70
B <sub>32</sub> W <sub>23</sub> C <sub>10</sub> V <sub>14</sub>	67, 68
B <sub>30</sub> W <sub>23</sub> C <sub>10</sub> E <sub>38</sub>	54 to 64
B <sub>31</sub> W <sub>23</sub> C <sub>10</sub> V <sub>14</sub> E <sub>38</sub>	15, 16
B <sub>31</sub> W <sub>23</sub> C <sub>10</sub> V <sub>14</sub> E <sub>38</sub> X <sub>6</sub>	17
B <sub>30</sub> W <sub>23</sub> C <sub>11</sub> V <sub>14</sub> E <sub>38</sub>	46 to 53
B <sub>30</sub> W <sub>23</sub> C <sub>10</sub> V <sub>14</sub> E <sub>38</sub>	2 to 14, 18 to 41, 43, 71 to 76
B <sub>30</sub> W <sub>23</sub> C <sub>10</sub> V <sub>14</sub> E <sub>38</sub> X <sub>5</sub>	1
B <sub>30</sub> W <sub>23</sub> C <sub>10</sub> V <sub>14</sub> E <sub>38</sub> X <sub>7</sub>	42

## TEST FACILITY

The General Dynamics/Convair low speed wind tunnel is a close-return atmospheric tunnel, constructed of reinforced concrete. Airflow is created by a 6-bladed propeller, 20 feet in diameter, powered by a 2250-horsepower synchronous motor.

The tunnel has a primary test section 8 feet high, 12 feet wide and 25 feet long. A model or component can be tested in this section at air-speeds up to 300 mph. The sides and ceiling of the test section are largely glass windows. These windows and fluorescent lights, housed in corner fillets in the test section, provide excellent model observation and model photography. Personnel access doors open directly into the test section from the control room and the model service area.

A variety of support systems are available for mounting a model or component. These support systems include single, bicycle, and three-strut supports, sting supports and supports for two dimensional installations.

Digital computer and plotting facilities are available so that the results of a test run can be processed, tabulated, and plotted within 10 minutes of the test run.

## DATA REDUCTION

Aerodynamic forces and moments were measured with the Convair C-5-1.35 six-component balance. Seven base and two cavity pressures were measured. The data presented have been corrected for tunnel wall effects before reducing to coefficient form about the Moment Reference Center (MRC). For all data presented, the MRC was at Fuselage Station (FS) 49.078 and Water Line (WL) 8.000, which corresponds to 29.078 inches from the model nose, and along the model centerline. Model wing dimensional data used to reduce the test data is given below for wing  $W_{23}$ :

$$S_{\text{ref}} = 492.48 \text{ sq. in.}$$

$$l_{\text{ref}} = 16.188 \text{ in}$$

$$b_{\text{ref}} = 34.632 \text{ in}$$

The total fuselage base area was  $68.565 \text{ in}^2$  and the fuselage cavity area was  $3.140 \text{ in}^2$  with the basic sting and  $8.293 \text{ in}^2$  with the wood sting sleeve.

The data was reduced to coefficient form using the normal expressions, except for a cavity pressure correction to the axial force coefficient. The two pressure tubes in the balance cavity were manifolded together to provide one measurement denoted as  $P_c$ . This pressure along with the average base pressure ( $P_b$ ) was used to correct the axial force as follows:

$$C_{A_{\text{TC}}} = \frac{(P_c - P_b)}{q} \quad \frac{A_c}{S_{\text{REF}}}$$

and  $C_A = C_{A_{\text{BALANCE}}} + C_{A_{\text{TC}}}$

where  $A_c$  = cavity area

TABLE I.

## TEST CONDITIONS

BALANCE UTILIZED: GD/Convair, C-5-1.35

**CAPACITY:**

### ACCURACY:

## COEFFICIENT TOLERANCE:

NF	<u>500 lbs</u>
SF	<u>500 lbs</u>
AF	<u>50 lbs</u>
PM	<u>1300 in-lbs</u>
YM	<u>1300 in-lbs</u>
RM	<u>240 in-lbs</u>

A set of five horizontal black lines spaced evenly apart.

A series of five horizontal black lines of varying lengths, starting from the top and decreasing in length downwards.

### **COMMENTS:**

TABLE II. TEST GDL5WT 603-0 DATA SET COLLATION SHEET

13

PRETEST  
 POSTTEST

DATA SET IDENTIFIER	CONFIGURATION	SCID.	PARAMETERS / VALUES					NO. OF RUNS	MACH NUMBERS (OR ALTERNATE INDEPENDENT VARIABLE)						
			a	b	c	d	e								
RD6 024	B <sub>30</sub> W <sub>23</sub> C <sub>10</sub> V <sub>14</sub> E <sub>38</sub>	A	0	0	+5	+5	0	1	24						
025		A	5	0	+5	+5	0	1	25						
026		A	0	0	+10	+10	0	1	26						
027		A	5	0	+10	+10	0	1	27						
028		A	0	0	-5	-5	0	1	28						
029		A	5	0	-5	-5	0	1	29						
030		A	0	0	-10	-10	0	1	30						
031		A	5	0	-10	-10	0	1	31						
032		A	0	0	+10	-10	0	1	32						
033		A	5	0	+10	-10	0	1	33						
042	B <sub>30</sub> W <sub>23</sub> C <sub>10</sub> V <sub>14</sub> E <sub>38</sub> X <sub>7</sub>	A	0	0	0	0	0	1	42						
043	B <sub>30</sub> W <sub>23</sub> C <sub>10</sub> V <sub>14</sub> E <sub>38</sub>	A	0	0	0	0	0	1	43						
044	B <sub>30</sub> W <sub>23</sub> V <sub>14</sub> E <sub>38</sub>	A	0	-	0	0	0	1	44						
045		A	5	-	0	0	0	1	45						
046	B <sub>30</sub> W <sub>23</sub> C <sub>11</sub> V <sub>14</sub> E <sub>38</sub>	A	0	0	0	0	0	1	46						
047		A	5	0	0	0	0	1	47						
048		A	0	10	0	0	0	1	48						
049		A	5	10	0	0	0	1	49						
050		A	0	-10	0	0	0	1	50						
051		A	5	-10	0	0	0	1	51						
			1	7	13	19	25	31	37	43	49	55	61	67	73,76

CL    CY    CP    CSL    CLM    CLN    CAF    CAB    CA    CN    IDPVAR(1) | IDPVAR(2) | NDV  
 COEFFICIENTS:  
 a or b       $\alpha A = -4^{\circ}$  to  $+24^{\circ}$  @  $\Delta x = 2^{\circ}$

SCHEDULES

TABLE II. TEST GDL5WT-603-0 DATA SET COLLATION SHEET (CONTINUED)

14

PRETEST  
 POSTTEST

DATA SET IDENTIFIER	CONFIGURATION	SCHD.		PARAMETERS/VALUES		NO. OF RUNS	MACH NUMBERS (OR ALTERNATE INDEPENDENT VARIABLE)
		$\alpha$	$\beta$	$\delta_C$	$\delta_{CL}$		
RD6001	$B_{30}W_{23}C_{10}V_{14}E_{38}X_5$	A	0	0	0	0	1
002	$B_{30}W_{23}C_{10}V_{14}E_{38}$	A	0	0	0	0	1
003		A	5	0	0	0	1
007		A	0	10	0	0	1
008		A	5	10	0	0	1
009		A	0	-10	0	0	1
010		A	5	-10	0	0	1
011		A	0	+15	0	0	1
012		A	5	+15	0	0	1
013		A	0	+20	0	0	1
014		A	5	+20	0	0	1
015	$B_{31}W_{23}C_{10}V_{14}E_{38}$	A	0	0	0	0	1
016		A	5	0	0	0	1
017	$B_{31}W_{23}C_{10}V_{14}E_{38}X_6$	A	0	0	0	0	1
018	$B_{30}W_{23}C_{10}V_{14}E_{38}$	A	0	-10	-10	0	1
019		A	5	-10	-10	0	1
020		A	0	+15	+10	0	1
021		A	5	+15	+10	0	1
022		A	0	+15	+5	0	1
023		A	5	+15	+5	0	1
						23	
							75.75
							10

CL CY CD GSL ICLM CLN CAF CAB CPA ICN  
 COEFFICIENTS: IDPVAR(1) IDPVAR(2) IDV

$\alpha$  or  $\beta$   
 SCHEDULES  
 $\Delta\alpha = -4$  to  $+24^\circ$  @  $\Delta\beta = 2^\circ$

TABLE II. TEST GDL5WT-603 DATA SET COLLATION SHEET (CONTINUED)

-5

PRETEST  
POSTTEST

COEFFICIENTS: CL CY CD CSL CLM CN CAF CAB CA CN MACH BETA 10

$$\text{SCHEDULES} \quad \text{a or b} \quad \beta_B = -620 + 10^\circ \quad (\text{a}) \quad \Delta\beta = 2^\circ$$

TABLE II. TEST GOLST 603-D DATA SET COLLATION SHEET (CONTINUED)

16

OPRETEST  
EPOSTTEST

DATA SET IDENTIFIER	CONFIGURATION	SCID.	PARAMETERS/VALUES										NO. OF RUNS	MACH NUMBERS
			"	"	SC	ET	SE	IN	SO	SERI	SE	KR	FERO	FR
RD6034	B30 W23 C10 V <sub>1/4</sub> E38	A	0	0	10	10	0	-10	-10	0	0	1	34	
		A	5	0	10	10	0	-10	-10	0	0	1	35	
		A	0	0	10	0	0	-10	0	0	0	1	36	
		A	5	0	10	0	0	-10	0	0	0	1	37	
		A	0	0	0	0	0	0	0	10	10	1	38	
		A	5	0	0	0	0	0	0	10	10	1	39	
		A	0	0	0	0	0	0	0	10	10	1	40	
		A	5	0	0	0	0	0	0	10	10	1	41	
		A	0	15	25	-10	-10	25	-10	-10	0	1	71	
		A	5	15	25	-10	-10	25	-10	-10	0	1	72	
		A	0	-10	-10	5	5	-10	5	5	0	1	73	
		A	5	-10	-10	5	5	-10	5	5	0	1	74	
		A	0	-10	-10	5	5	-10	5	5	0	1	75	
		A	5	-10	-10	5	5	-10	5	5	0	1	76	
		A	0	-10	-10	5	5	-10	5	5	0	1	77	
		A	5	-10	-10	5	5	-10	5	5	0	1	78	
		A	0	-10	-10	5	5	-10	5	5	0	1	79	
		A	5	-10	-10	5	5	-10	5	5	0	1	80	
		A	0	-10	-10	5	5	-10	5	5	0	1	81	
		A	5	-10	-10	5	5	-10	5	5	0	1	82	
		A	0	-10	-10	5	5	-10	5	5	0	1	83	
		A	5	-10	-10	5	5	-10	5	5	0	1	84	
		A	0	-10	-10	5	5	-10	5	5	0	1	85	
		A	5	-10	-10	5	5	-10	5	5	0	1	86	
		A	0	-10	-10	5	5	-10	5	5	0	1	87	
		A	5	-10	-10	5	5	-10	5	5	0	1	88	
		A	0	-10	-10	5	5	-10	5	5	0	1	89	
		A	5	-10	-10	5	5	-10	5	5	0	1	90	
		A	0	-10	-10	5	5	-10	5	5	0	1	91	
		A	5	-10	-10	5	5	-10	5	5	0	1	92	
		A	0	-10	-10	5	5	-10	5	5	0	1	93	
		A	5	-10	-10	5	5	-10	5	5	0	1	94	
		A	0	-10	-10	5	5	-10	5	5	0	1	95	
		A	5	-10	-10	5	5	-10	5	5	0	1	96	
		A	0	-10	-10	5	5	-10	5	5	0	1	97	
		A	5	-10	-10	5	5	-10	5	5	0	1	98	
		A	0	-10	-10	5	5	-10	5	5	0	1	99	
		A	5	-10	-10	5	5	-10	5	5	0	1	100	

CONFIDENTIALS: C7 CK CD CSC CWM CLN CAF CAB CA CN [REDACTED] [REDACTED] 10

$$\alpha = -4^{\circ} \text{ to } +24^{\circ} \text{ @ } \Delta\alpha = 2^{\circ}$$

TABLE II. TEST GDLST 603-0 DATA SET COLLATION SHEET (CONTINUED)

17

PRETEST  
 POSTTEST

DATA SET IDENTIFIER	CONFIGURATION	SCHD.	PARAMETERS/VALUES					NO. OF RUNS	MACH NUMBERS (OR ALTERNATE INDEPENDENT VARIABLE)	
			a	b	c	d	e			
RD6052	$B_{30} W_{23} C_{10} V_{14} E_{38}$	A	0	15	0	0	0	1	52	
053	$B_{30} W_{23} C_{10} E_{38}$	A	5	15	0	0	0	1	53	
054	$B_{30} W_{23} C_{10} E_{38}$	A	0	0	0	0	-	1	54	
055		A	5	0	0	0	-	1	55	
059		A	0	-10	0	0	-	1	59	
060		A	5	-10	0	0	-	1	60	
061		A	0	+10	0	0	-	1	61	
062		A	5	+10	0	0	-	1	62	
063		A	0	+15	0	0	-	1	63	
064		A	5	+15	0	0	-	1	64	
065	$B_{30} W_{23} E_{38}$	A	0	-	0	0	-	1	65	
066		A	5	-	0	0	-	1	66	
067	$B_{30} W_{23} C_{10} V_{14}$	A	0	0	0	0	0	1	67	
068		A	5	0	0	0	0	1	68	
069	$B_{30} W_{23} C_{10} V_{14}$	A	0	0	0	0	0	1	69	
070		A	5	0	0	0	0	1	70	
075	$B_{30} W_{23} C_{10} V_{14} E_{38}$	A	0	0	0	0	0	1	75	
076		A	0	0	0	0	0	1	76	
077	$B_{30} W_{23} V_{14}$	A	0	0	0	0	0	1	77	
078		A	5	0	0	0	0	1	78	
			7	13	19	25	31	37	43	49
									55	61
									67	75
										76
										10

MODEL 4 INVERTED  
 MODEL UPRIGHT - REPEAT RUN

CL. ICY CP. ICSL ICLM CLM CAF CAB CA CN.  
 COEFFICIENTS:  
 $\alpha$  or  $\beta$   
 SCHEDULES

IDPVAR(1)|IDPVAR(2)|NDV

$\alpha_A = -4$  to  $+24^\circ$  @  $\Delta\alpha = 2^\circ$

TABLE II. TEST GDLST 603-0 DATA SET COLLATION SHEET (CONTINUED)

四庫全書

POSTTEST

DATA SET IDENTIFIER	CONFIGURATION	SCMD.				PARAMETERS / VALUES NO. of RUNS	MACH NUMBERS (OR ALTERNATE INDEPENDENT VARIABLE)	
		a	b	Sc	Seu	See	SP	
RD6079	B30 W23 V16	A	0	-	0	0	1	79
080		A	5	-	0	0	1	80
081	B30 W23 C10 V16	A	0	0	0	0	1	81
082		A	5	0	0	0	1	82
083	B30 C10	A	0	0	-	-	1	83
084		A	5	0	-	-	1	84
085		A	0	-10	-	-	1	85
086		A	5	-10	-	-	1	86
087		A	0	+10	-	-	1	87
088		A	5	+10	-	-	1	88
089		A	0	+15	-	-	1	89
090		A	5	+15	-	-	1	90
091	B30 V4	A	0	-	-	0	1	91
092		A	5	-	-	0	1	92
093	B30	A	0	-	-	-	1	93
094		A	5	-	-	-	1	94
095	B32	A	0	-	-	-	1	95
096		A	5	-	-	-	1	96

	$C_L$	$C_Y$	$C_D$	$C_{SL}$	$C_{LM}$	$C_{LN}$	$C_{AF}$	$C_{AB}$	$C_A$	$C_N$	$ IDPVAR(1) $	$ IDPVAR(2) $	$ NDV $
COEFFICIENTS:	1	7	13	19	25	31	37	43	49	55	61	67	73

$$\alpha = -4^{\circ} \text{ to } +24^{\circ} \quad \Delta\alpha = 2^{\circ}$$

TABLE III. DIMENSIONAL DATA

MODEL COMPONENT: B<sub>30</sub>GENERAL DESCRIPTION: Basic B-18E3 booster configuration with R<sub>5</sub> rocket engines mounted, canard fairings, canopy and rocket engines shrouds.DRAWING NUMBER: WT-71-107502

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length (to wing T.E. @ B.L. 0.0)	<u>173.33 ft</u>	<u>41.60 in</u>
Max. Width	<u>33.00 ft</u>	<u>7.92 in</u>
Max. Depth	<u>33.00 ft</u>	<u>7.92 in</u>
Fineness Ratio	<u>5.253</u>	<u>5.253</u>
Area		
Max. Cross-Sectional	<u>855.29 ft<sup>2</sup></u>	<u>49.265 in<sup>2</sup></u>
Planform	<u>                </u>	<u>                </u>
Wetted	<u>                </u>	<u>                </u>
Base	<u>                </u>	<u>                </u>
Length (to model base area)	<u>159.79 ft</u>	<u>38.35 in</u>
Fineness Ratio (to model base area)	<u>4.842</u>	<u>4.842</u>

TABLE III. (CONTINUED).

MODEL COMPONENT: B<sub>31</sub>GENERAL DESCRIPTION: Same as body B<sub>30</sub>, but with a blunter nose (larger nose radius).DRAWING NUMBER: WT-71-107511

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length (to wing T.E. @ B.L. 0.0)	<u>169.0 ft</u>	<u>40.560 in</u>
Max. Width	<u>33.00 ft</u>	<u>7.92 in</u>
Max. Depth	<u>33.00 ft</u>	<u>7.92 in</u>
Fineness Ratio	<u>5.121</u>	<u>5.121</u>
Area		
Max. Cross-Sectional	<u>855.29 ft<sup>2</sup></u>	<u>49.265 in<sup>2</sup></u>
Planform		
Wetted		
Base		
Length (to model base area)	<u>155.46 ft</u>	<u>37.310 in</u>
Fineness Ratio (to model base area)	<u>4.711</u>	<u>4.711</u>

TABLE III. (CONTINUED)

MODEL COMPONENT: BODY - B<sub>32</sub>GENERAL DESCRIPTION: Same as body B<sub>30</sub>, but with a full scale 8 foot length section removed from the fuselage.DRAWING NUMBER: WT-71-107502

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length (to wing T.E. @ B.L.O.O)	<u>165.33 ft</u>	<u>39.679 in</u>
Max. Width	<u>33.00 ft</u>	<u>7.92 in</u>
Max. Depth	<u>33.00 ft</u>	<u>7.92 in</u>
Fineness Ratio	<u>5.010</u>	<u>5.010</u>
Area		
Max. Cross-Sectional	<u>855.29 ft<sup>2</sup></u>	<u>49.265 in<sup>2</sup></u>
Planform		
Wetted		
Base		
Length (to model base area)	<u>151.79 ft</u>	<u>36.430 in</u>
Fineness Ratio (to model base area)	<u>4.600</u>	<u>4.600</u>

TABLE III. (CONTINUED)

<u>MODEL COMPONENT:</u>	<u>Wing W<sub>23</sub></u>
<u>GENERAL DESCRIPTION:</u>	Basic B-18E3 wing with full scale total characteristics of 8549 ft <sup>2</sup> , A=2.436, = 0.2158, L.E. sweep = 53°, and spanwise split elevons.
	Has three airfoil thickness

DRAWING NUMBER: WT-71-107503

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
<u>TOTAL DATA</u>		
<u>Area</u>		
Planform	<u>8549 ft<sup>2</sup></u>	<u>492.43 in<sup>2</sup></u>
Wetted		
Span (equivalent)	<u>144.3 ft</u>	<u>34.634 in</u>
Aspect Ratio	<u>2.436</u>	<u>2.436</u>
Rate of Taper		
Taper Ratio	<u>0.2158</u>	<u>0.2158</u>
Dihedral Angle, degrees	<u>4°</u>	<u>4°</u>
Incidence Angle, degrees	<u>2°</u>	<u>2°</u>
Aerodynamic Twist, degrees	<u>0°</u>	<u>0°</u>
Toe-In Angle		
Cant Angle		
Sweep Back Angles, degrees		
Leading Edge	<u>53°</u>	<u>53°</u>
Trailing Edge	<u>15°</u>	<u>15°</u>
0.25 Element Line	<u>46.5°</u>	<u>46.5°</u>
Chords:		
Root (Wing Sta. 0.0)	<u>97.45 ft</u>	<u>23.388 in</u>
Tip, (equivalent)	<u>21.03 ft</u>	<u>5.048 in</u>
MAC	<u>67.46 ft</u>	<u>16.19 in</u>
Fus. Sta. of .25 MAC	<u>209.0 ft</u>	<u>50.166 in</u>
W.L. of .25 MAC	<u>23.65 ft</u>	<u>5.676 in</u>
B.L. of .25 MAC	<u>28.25 ft</u>	<u>6.780 in</u>
Airfoil Section		
Root	<u>NACA 0008-64</u>	<u>(MOD)</u>
Tip	<u>NACA 0010-64</u>	
<u>EXPOSED DATA</u>		
<u>Area</u>		
Span, (equivalent)	<u>5621.5 ft<sup>2</sup></u>	<u>323.8 in<sup>2</sup></u>
Aspect Ratio	<u>111.31 ft</u>	<u>26.714 in</u>
Taper Ratio	<u>2.204</u>	<u>2.204</u>
Chords		
Root	<u>79.98 ft</u>	<u>19.194 in</u>
Tip	<u>21.03 ft</u>	<u>5.048 in</u>
MAC	<u>56.23 ft</u>	<u>13.497</u>
Fus. Sta. of .25 MAC	<u>220.29 ft</u>	<u>52.868 in</u>
W.L. of .25 MAC	<u>24.00 ft</u>	<u>5.760 in</u>
B.L. of .25 MAC	<u>38.82 ft</u>	<u>9.317 in</u>

TABLE III. (CONTINUED)

MODEL COMPONENT: Canard C<sub>10</sub>GENERAL DESCRIPTION: Basic delta canard for the B-18E3 with 485 ft<sup>2</sup> full scale planform area (both canards).DRAWING NUMBER:

DIMENSIONS - EXPOSED: per canard panel

	FULL SCALE	MODEL SCALE
Area	242.50 ft <sup>2</sup>	13.968 in <sup>2</sup>
Span (equivalent)	37.05 ft	8.892 in
Aspect Ratio	2.830	2.830
Taper Ratio	0.1698	0.1698
Dihedral Angle	0°	0°
Incidence Angle	0°	0°
Sweep Back Angles		
Leading Edge	45°	45°
Trailing Edge	0°	0°
0.25 Chord Line	36.83°	36.83°
Chords		
Root perpendicular to trailing edge	22.38 ft	5.371 in
Tip	3.80 ft	0.912 in
MAC	15.29 ft	3.669 in
Airfoil Section		
Root	NACA 0010-64	
Tip	NACA 0010-64	
Fuse. Sta. of .25 MAC	111.25 ft	26.70 in
W.L. of .25 MAC	38.33 ft	9.20 in
B.L. of .25 MAC	21.57 ft	5.176 in
Pivot Location		
Percent root chord	50%	50.0%
Fuselage station	111.48 ft	26.754 in
Water line	38.33 ft	9.20 in

TABLE III. (CONTINUED)

MODEL COMPONENT:	Canard C11
GENERAL DESCRIPTION:	Alternate canard with a $364 \text{ ft}^2$ full scale planform area (both canards). It represents 75% of canard C10 area

DRAWING NUMBER:

DIMENSIONS - EXPOSED: per canard	FULL SCALE	MODEL SCALE
Area	<u>181.86 ft<sup>2</sup></u>	<u>10.475 in<sup>2</sup></u>
Span (equivalent)	<u>32.09 ft</u>	<u>7.701 in</u>
Aspect Ratio	<u>2.830</u>	<u>2.830</u>
Taper Ratio	<u>0.1700</u>	<u>0.170</u>
Dihedral Angle	<u>0°</u>	<u>0°</u>
Incidence Angle	<u>0°</u>	<u>0°</u>
Sweep Back Angles		
Leading Edge	<u>45°</u>	<u>45°</u>
Trailing Edge	<u>0°</u>	<u>0°</u>
0.25 Chord Line	<u>36.83°</u>	<u>36.83°</u>
Chords		
Root	<u>19.38 ft</u>	<u>4.651 in</u>
Tip	<u>3.29 ft</u>	<u>0.790 in</u>
MAC	<u>13.24 ft</u>	<u>3.177 in</u>
Airfoil Section		
Root	<u>NACA 0010-64</u>	
Tip	<u>NACA 0010-64</u>	
Fuse. Sta. of .25 MAC	<u>111.25 ft</u>	<u>26.70 in</u>
W.L. of .25 MAC	<u>38.33 ft</u>	<u>9.20 in</u>
B.L. of .25 MAC	<u>20.59 ft</u>	<u>4.941 in</u>
Pivot Location		
Percent root chord	<u>50%</u>	<u>50%</u>
Fuselage station	<u>111.46 ft</u>	<u>26.754 in</u>
Water line	<u>38.33 ft</u>	<u>9.20 in</u>

TABLE III. (CONTINUED)

MODEL COMPONENT: Vertical Tail V<sub>14</sub>GENERAL DESCRIPTION: Basic tail for the B-18E3 configuration, 1300 ft<sup>2</sup>

full scale area

DRAWING NUMBER:

WT-71-107509 sheet 1

DIMENSIONS - EXPOSED:

	FULL SCALE	MODEL SCALE
Area	<u>1299.82 ft<sup>2</sup></u>	<u>74.870 in<sup>2</sup></u>
Span (equivalent)	<u>42.66 ft</u>	<u>10.238 in</u>
Aspect Ratio	<u>1.40</u>	<u>1.40</u>
Taper Ratio	<u>0.40</u>	<u>0.400</u>
Dihedral Angle	<u>-</u>	<u>-</u>
Incidence Angle	<u>-</u>	<u>-</u>
Sweep Back Angles		
Leading Edge	<u>40°</u>	<u>40°</u>
Trailing Edge	<u>13°</u>	<u>13°</u>
Hingeline	<u>24°</u>	<u>-</u>
Chords		
Root	<u>43.53 ft</u>	<u>10.448 in</u>
Tip	<u>17.41 ft</u>	<u>4.178 in</u>
MAC	<u>32.34 ft</u>	<u>7.761 in</u>
Airfoil Section		
Root	<u>NACA 0013-64</u>	<u>-</u>
Tip	<u>NACA 0011-64</u>	<u>-</u>
Fuse. Sta. of .25 MAC	<u>236.67 ft</u>	<u>56.800 in</u>
W.L. of .25 MAC	<u>68.25 ft</u>	<u>16.379 in</u>
B.L. of .25 MAC	<u>0.0</u>	<u>0.0 in</u>
B-18E3 Ref. Locations		
Fuse. Sta. of RMC	<u>204.49 ft</u>	<u>49.078 in</u>
W.L. of RMC	<u>33.33 ft</u>	<u>8.0 in</u>
W.L. of E	<u>33.33 ft</u>	<u>8.0 in</u>
B.L. of RMC	<u>0.0</u>	<u>0.0</u>

TABLE III. (CONCLUDED)

MODEL COMPONENT: Vertical Tail V16

GENERAL DESCRIPTION: Vertical tail for the B-18E2 Configuration, 2200 ft<sup>2</sup>  
full scale area.

DRAWING NUMBER: WT-71-107509 sheet 2

## DIMENSIONS - EXPOSED:

	FULL SCALE	MODEL SCALE
Area	2210. ft <sup>2</sup>	127.30 in <sup>2</sup>
Span (equivalent)	55.60 ft	13.344 in
Aspect Ratio	1.40	1.40
Taper Ratio	0.4247	0.4247
Dihedral Angle		
Incidence Angle		
Sweep Back Angles		
Leading Edge	40°	40°
Trailing Edge	15°	15°
	25°	25°
Chords		
Root	55.80 ft	13.392 in
Tip	23.70 ft	5.688 in
MAC	42.33 ft	10.059 in
Airfoil Section		
Root	NACA 0013-64	
Tip	NACA 0011-64	
Fuse. Sta. of .25 MAC	232.38 ft	55.770 in
W.L. of .25 MAC	74.15 ft	17.797 in
B.L. of .25 MAC	0.0 ft	0.0 in
B-18E2 Ref: Locations		
Fuse. Sta. of RMC	200.33 ft	48.078 in
W.L. of RMC	33.33 ft	8.0 in
W.L. of Q	33.33 ft	8.0 in
B.L. of RMC	0.0	0.0
Fuse. Sta. of Nose	79.17 ft	19.0 in

TABLE IV. INDEX OF MODEL FIGURES

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Effect of Full Span Elevon Deflection on Longitudinal Characteristics of the Basic B-18E3 Configuration, CANARD = 0	A	ELEVTR	10-11
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Effect of Nose Blunting, B30 vs. B31	B	Configuration, BETA	20-22
Effect of Canard Size, C10 vs. C11 and Canard Off, Canard Deflection = 0	B	Configuration, BETA	23-25
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Effect of Canard Size, C10 vs. C11 and Canard Off, Canard Deflection = 10 Degrees	B	Configuration, BETA	29-31

TABLE V. INDEX OF DATA FIGURES  
(Continued)

TITLE	PIOTTED COEFFICIENTS	CONDITIONS VARYING	PAGES
SCHEDULE			
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TABLE V. INDEX OF DATA FIGURES  
(Concluded)

PLOTTED COEFFICIENTS SCHEDULE:

- (A) CL vs. ALPHA, CLM, CD, L/D
- (B) CL vs. ALPHA, CLM, CD, L/D  
CLN, CY, CSL vs. ALPHA
- (C) CD, CAB, L/D vs. ALPHA
- (D) CLN, CY, CSL vs. BETA

MODEL FIGURES

Notes:

- Positive directions of force coefficients moment coefficients, and angles are indicated by arrows.

- For clarity, origins of wind and stability axes have been displaced from the center of gravity.

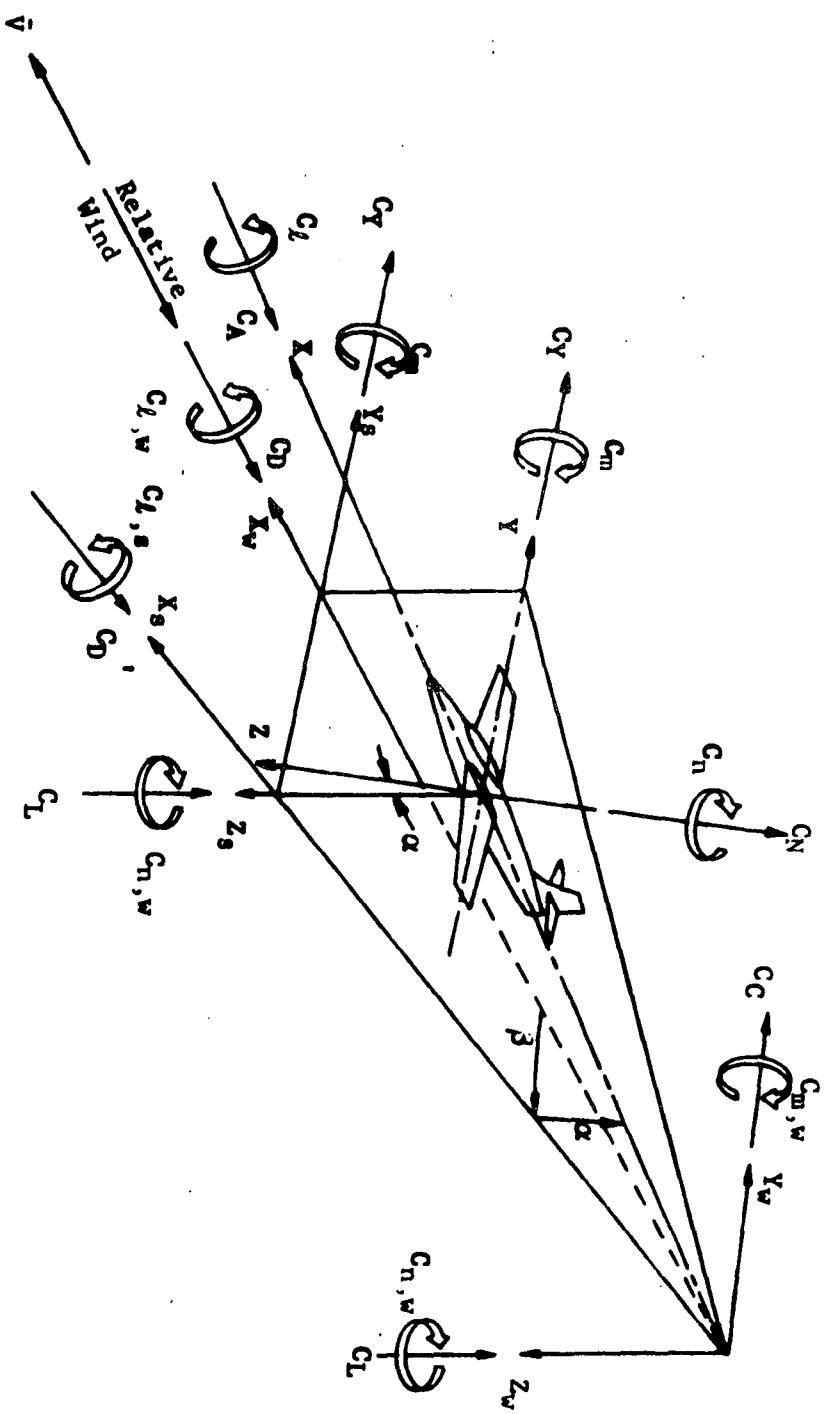


Figure 1. Axis systems, showing direction and sense of force and moment coefficients, angle of attack, and sideslip angle

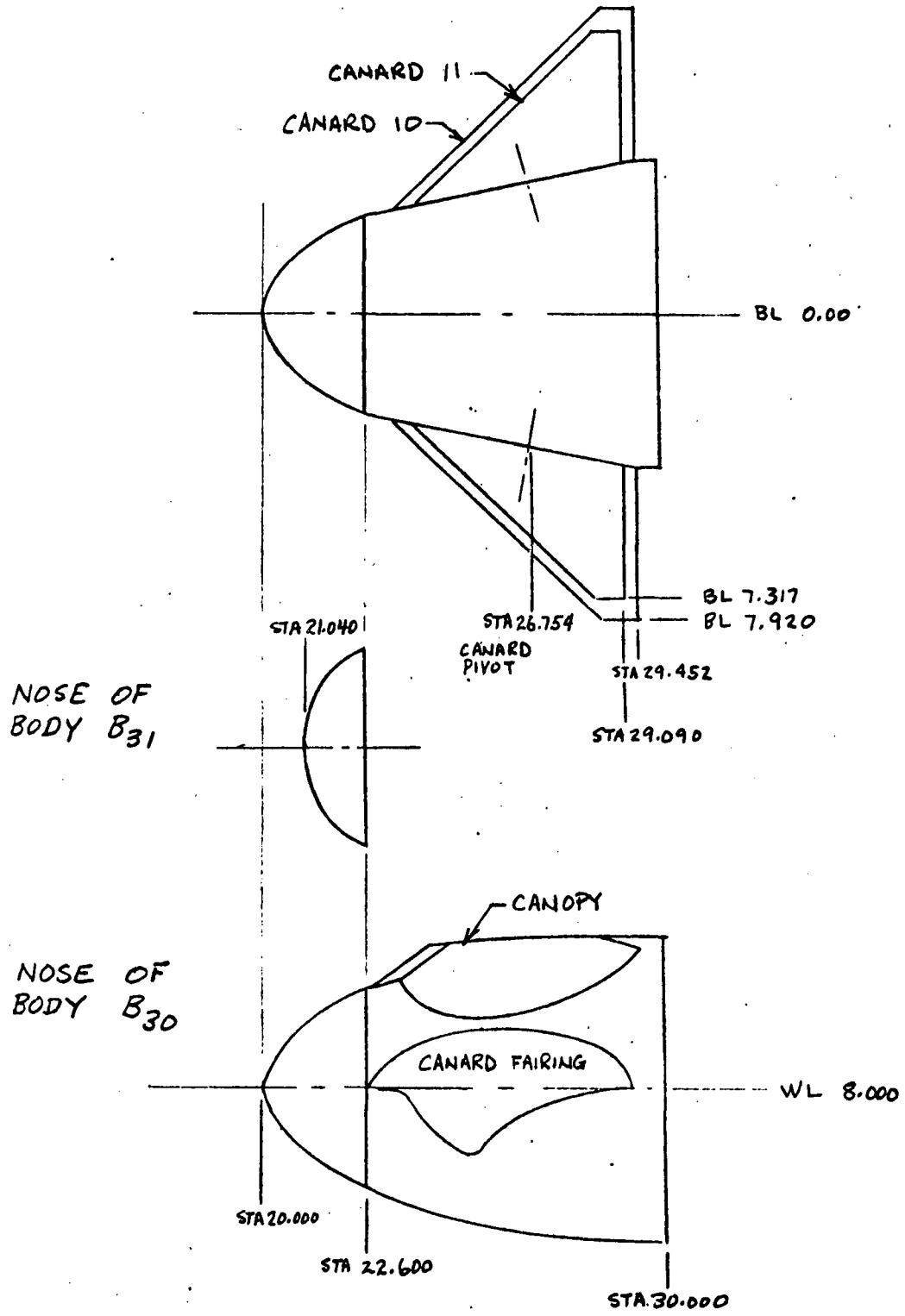


Figure 2. Nose of Body  $B_{30}$  and  $B_{31}$

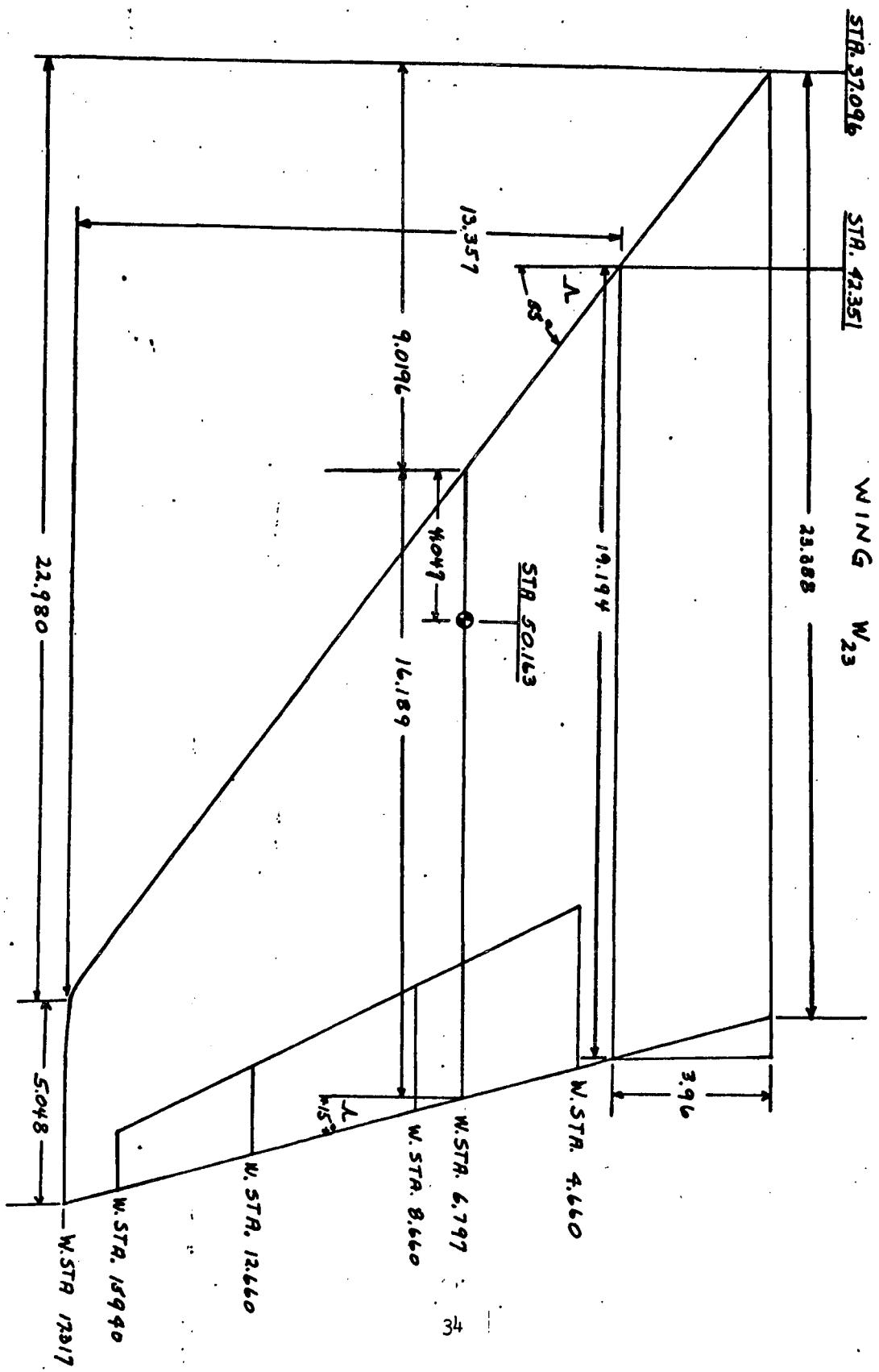


Figure 3. Wing  $W_{23}$

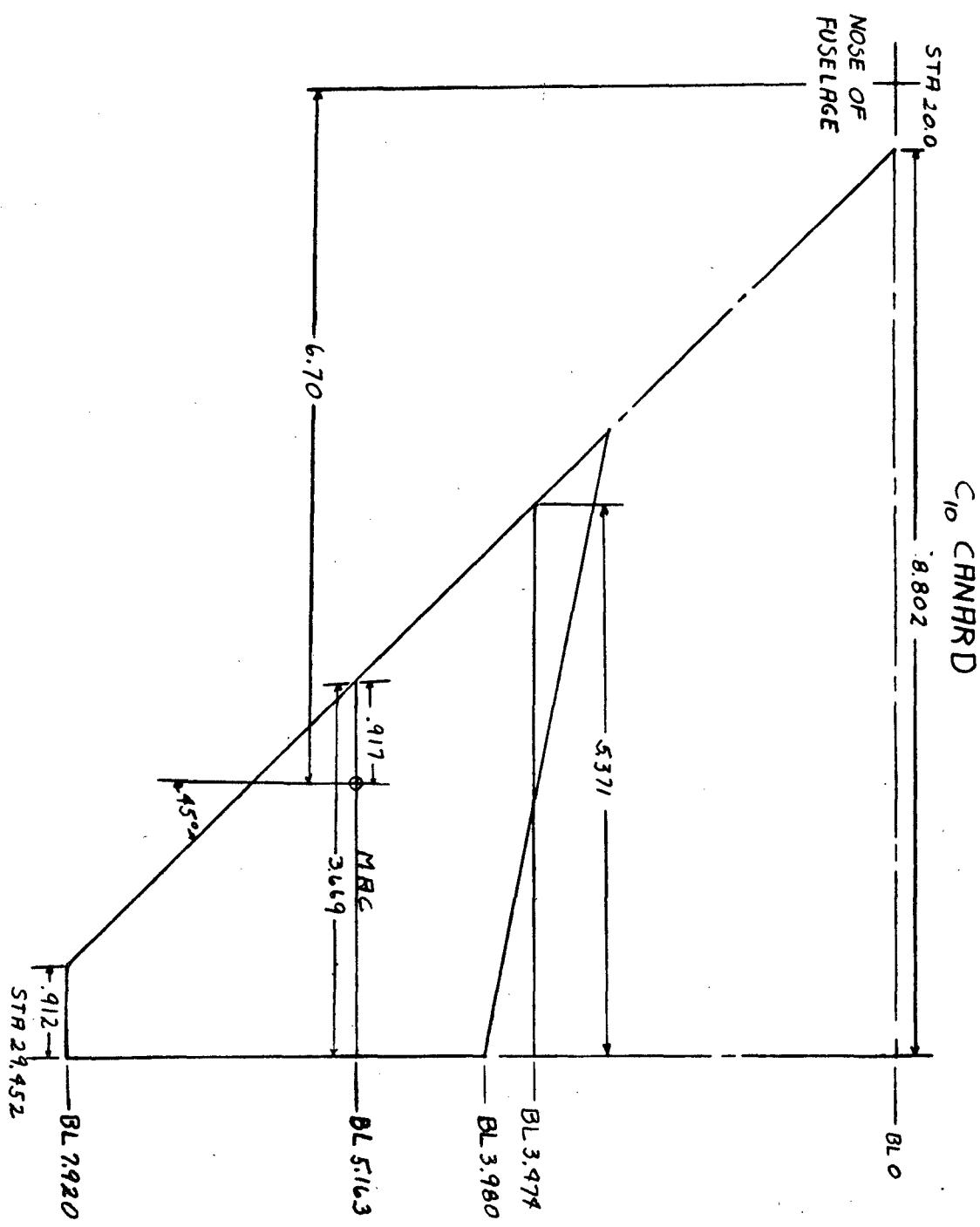


Figure 4. Canard C10

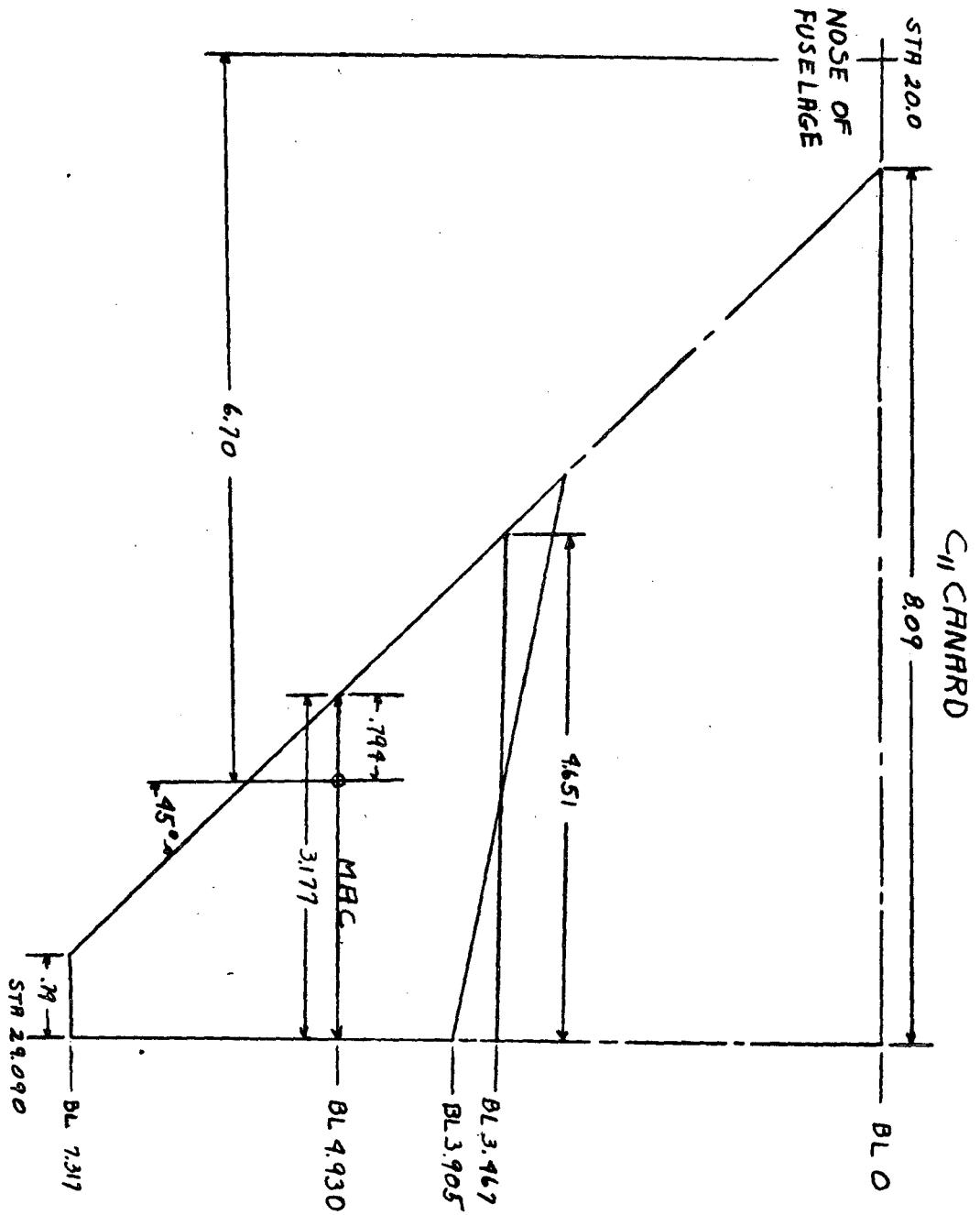


Figure 5. Canard C<sub>11</sub>

VERTICAL TAIL  $V_{14}$

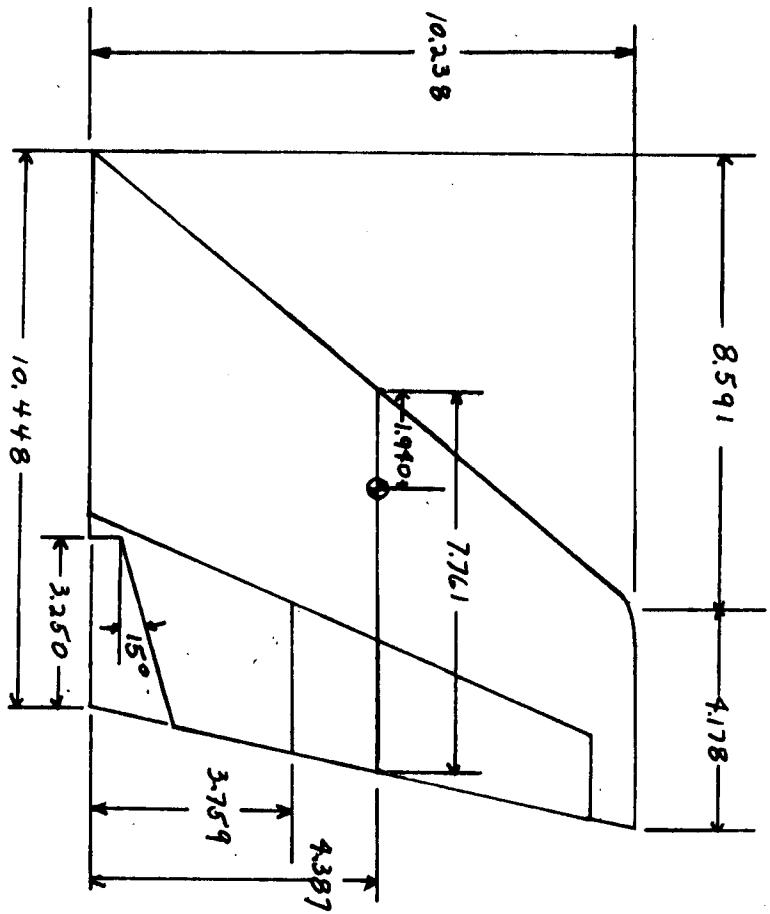


Figure 6. Vertical Tail  $V_{14}$

VERTICAL TAIL V<sub>16</sub>

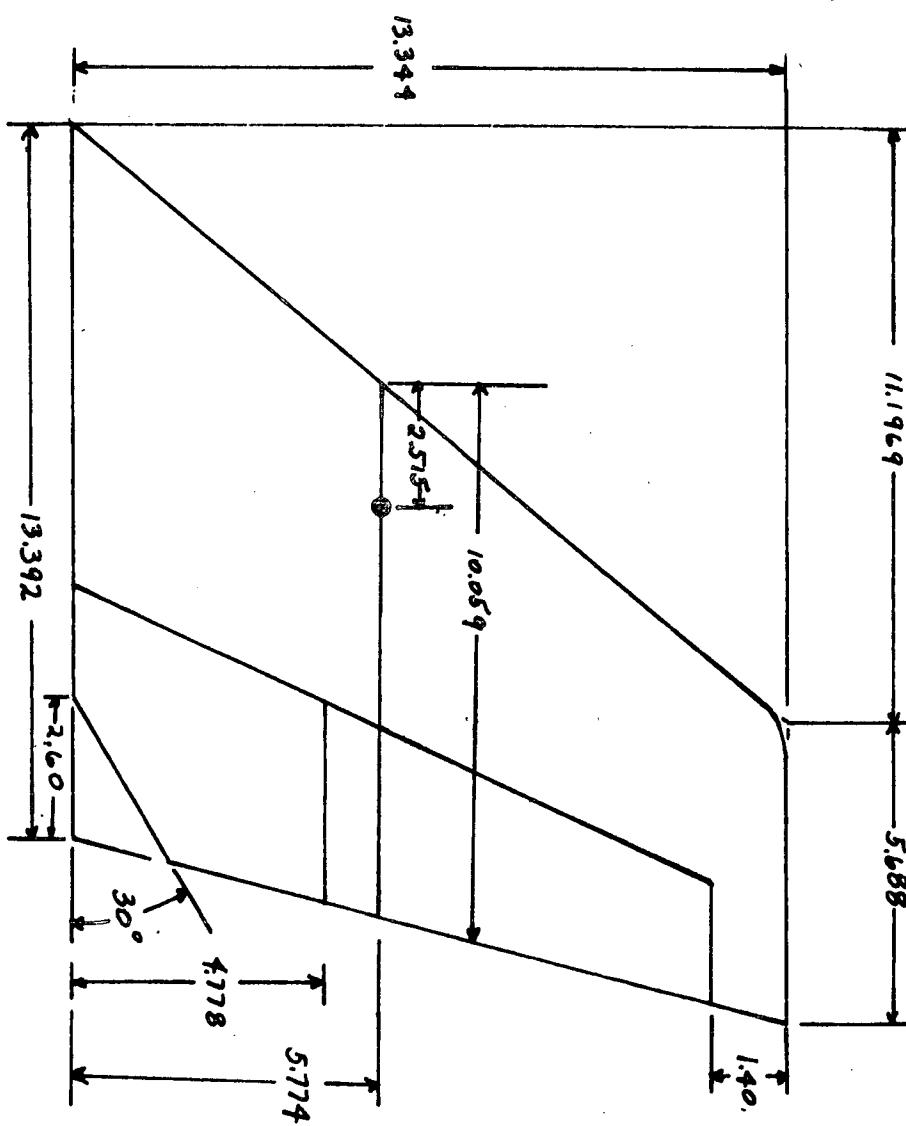


Figure 7. Vertical Tail V<sub>16</sub>

TAP LOCATIONS

STA 58.350    61.640

$P_{b_1}$

$P_{b_2}$

$P_{b_3}$

$P_{b_4}$

$P_{b_5}$

$P_{b_6}$

$P_{b_7}$

$P_{b_8}$

MANIFOLDED  
TOGETHER

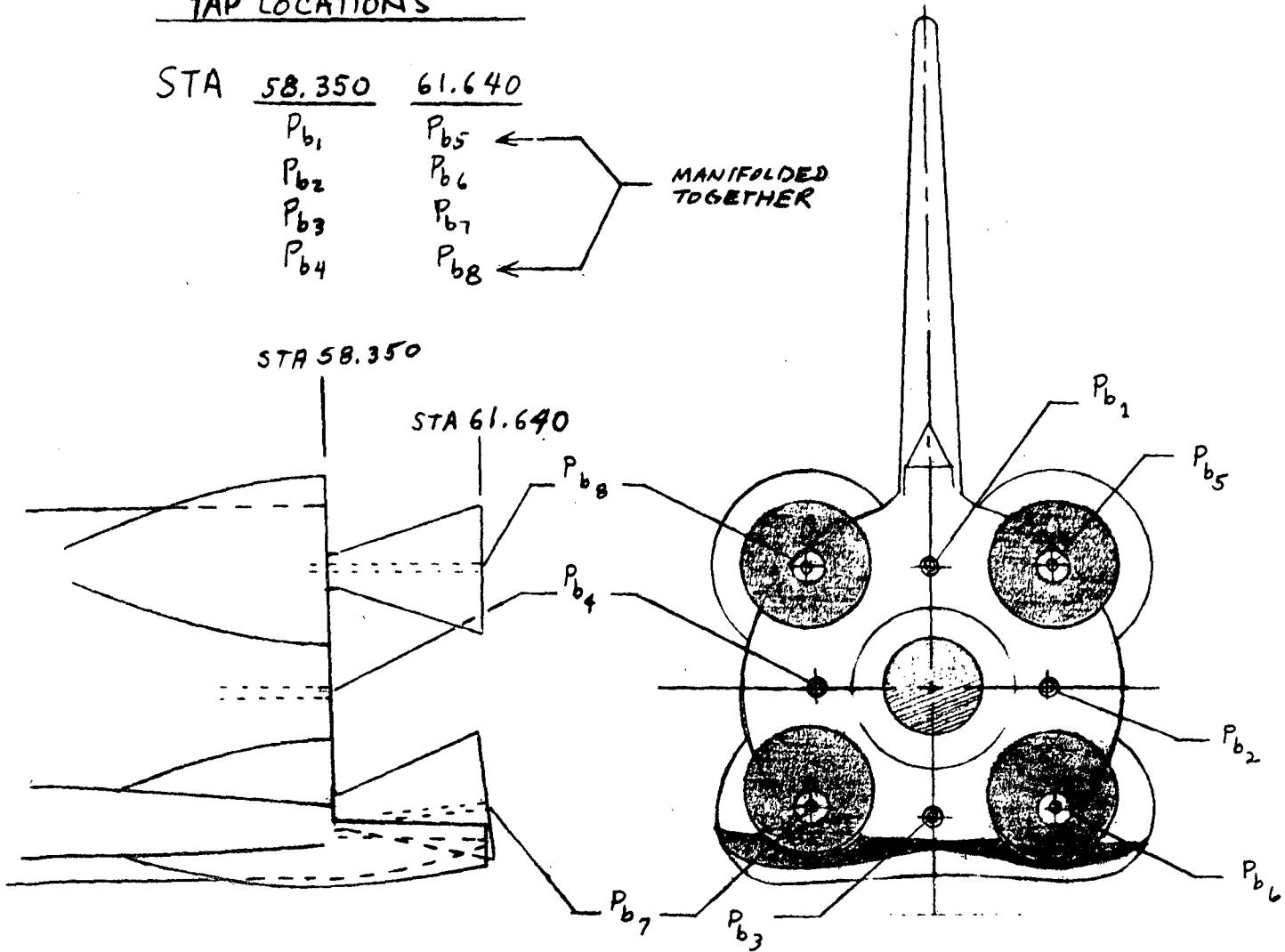


Figure 8. BASE REGION AND PRESSURE TAP LOCATIONS

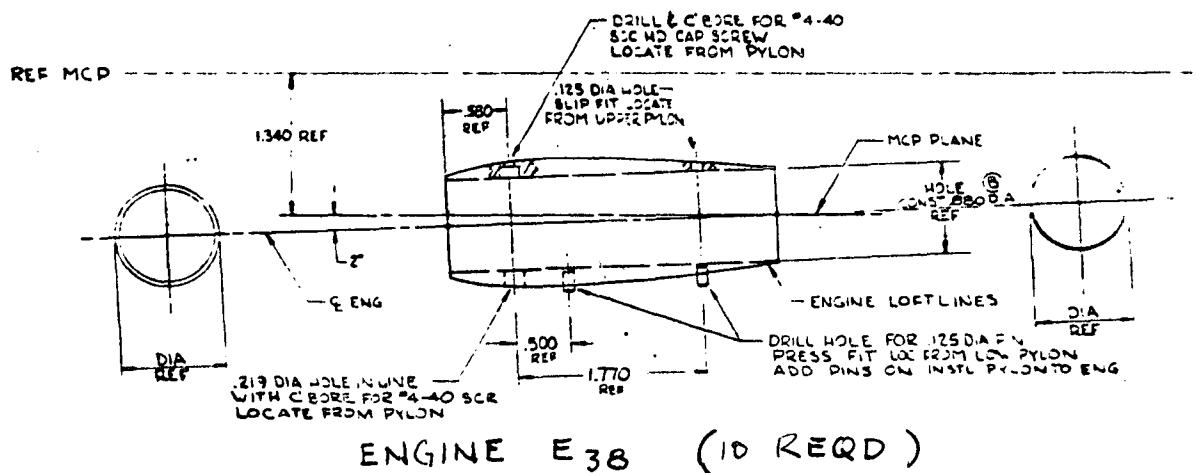
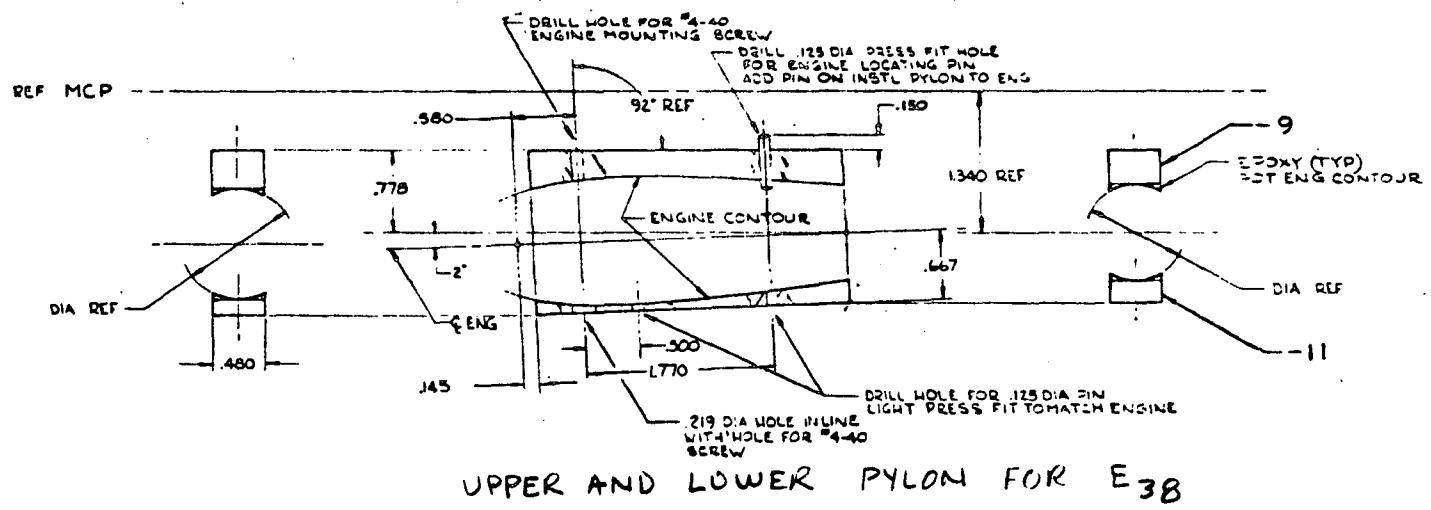
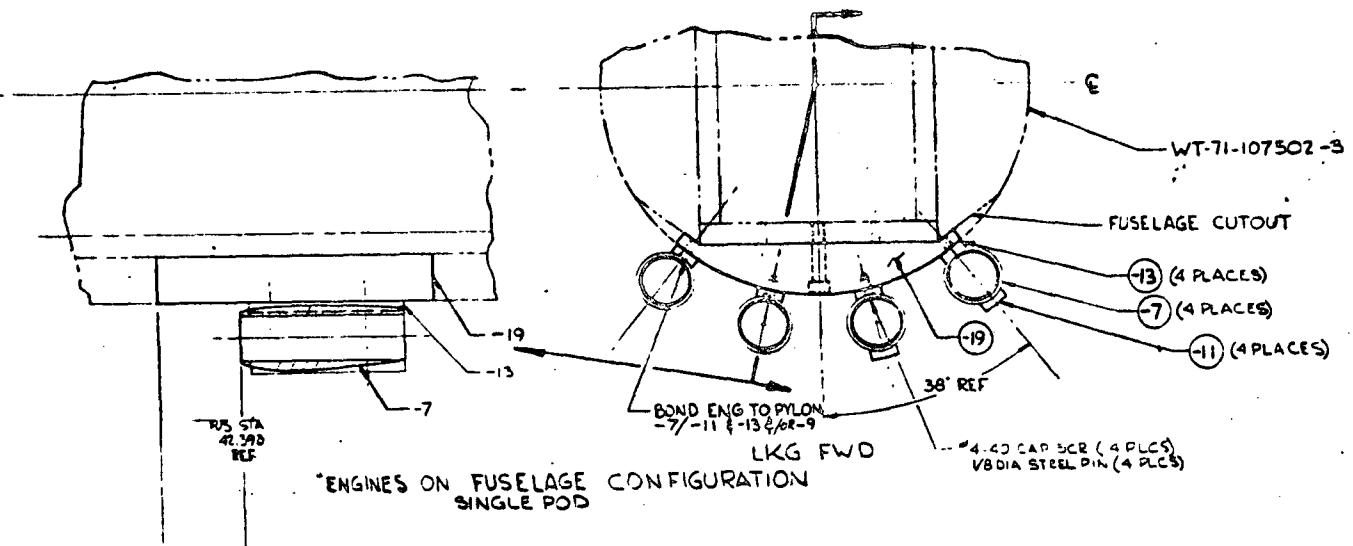


Figure 9. Engine E38

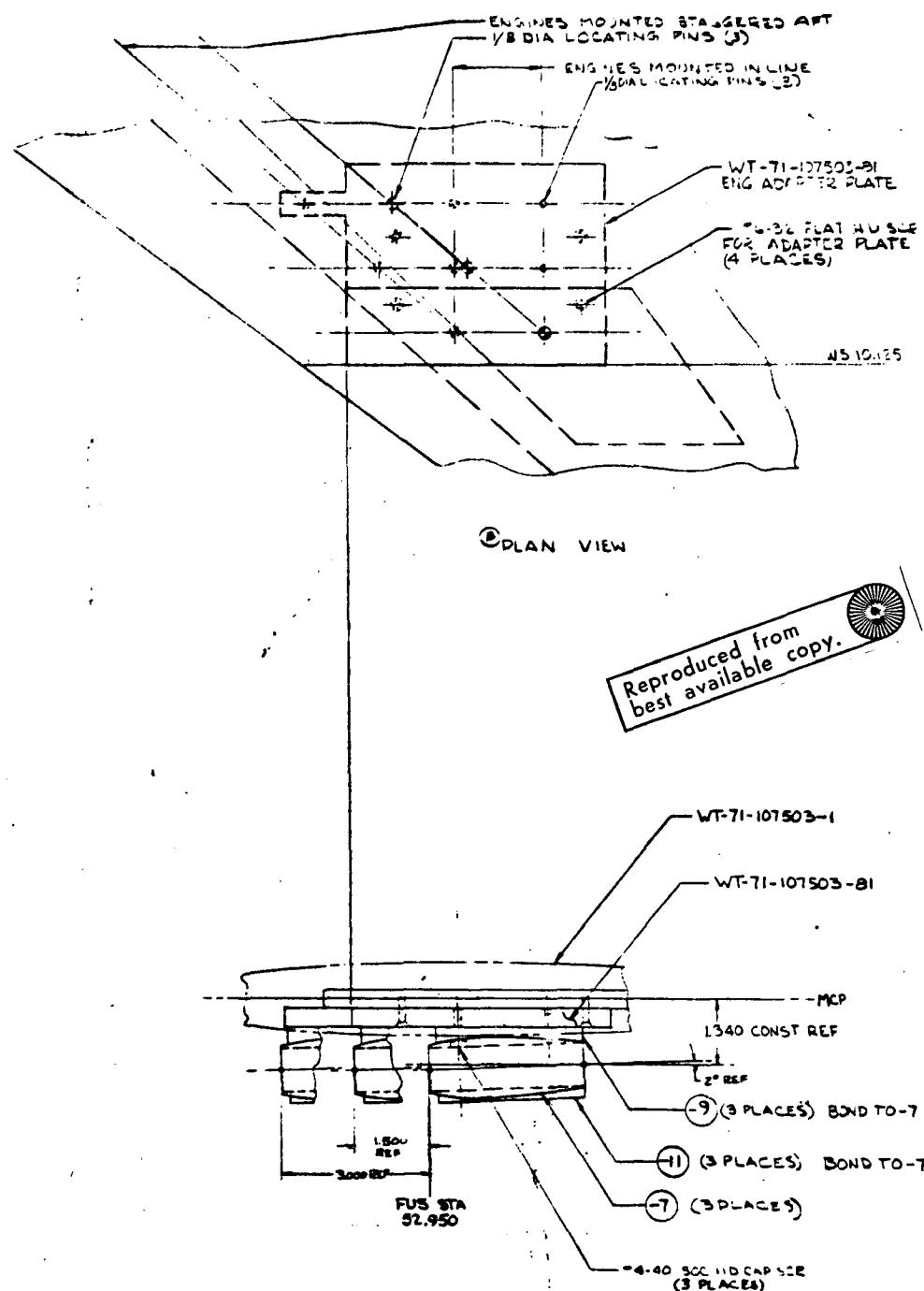


Figure 10. ENGINES ON WING

ROCKET ENGINES R<sub>5</sub>

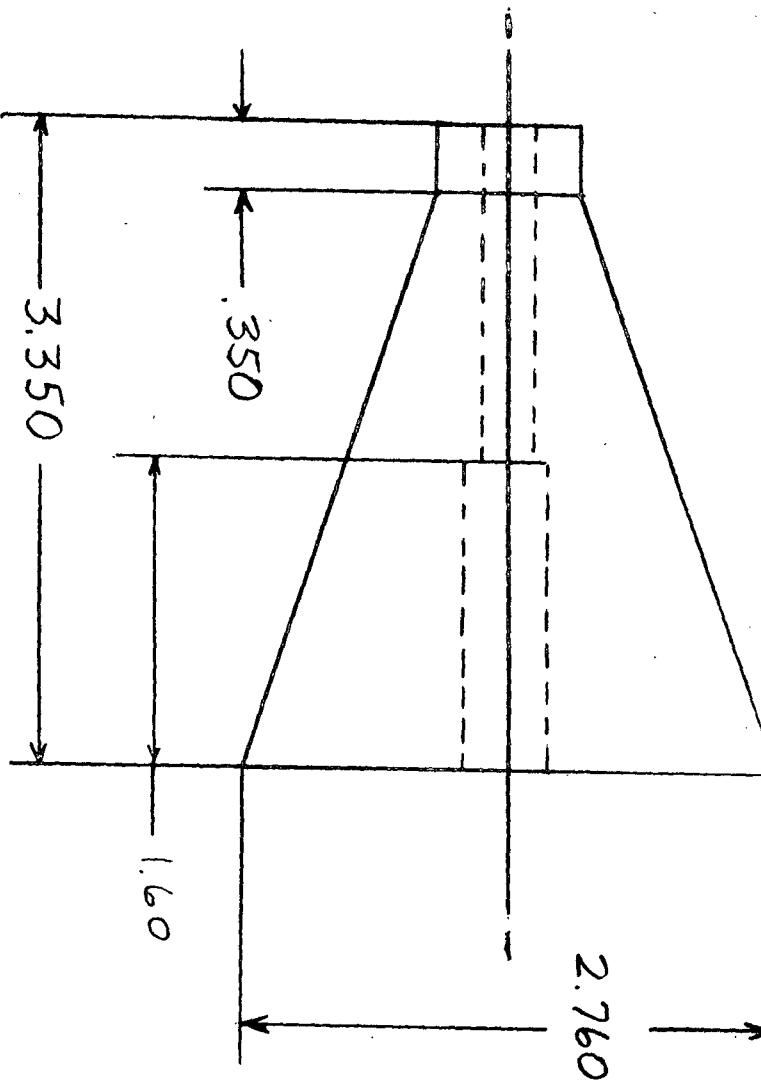


Figure 11. Rocket Engine R<sub>5</sub>

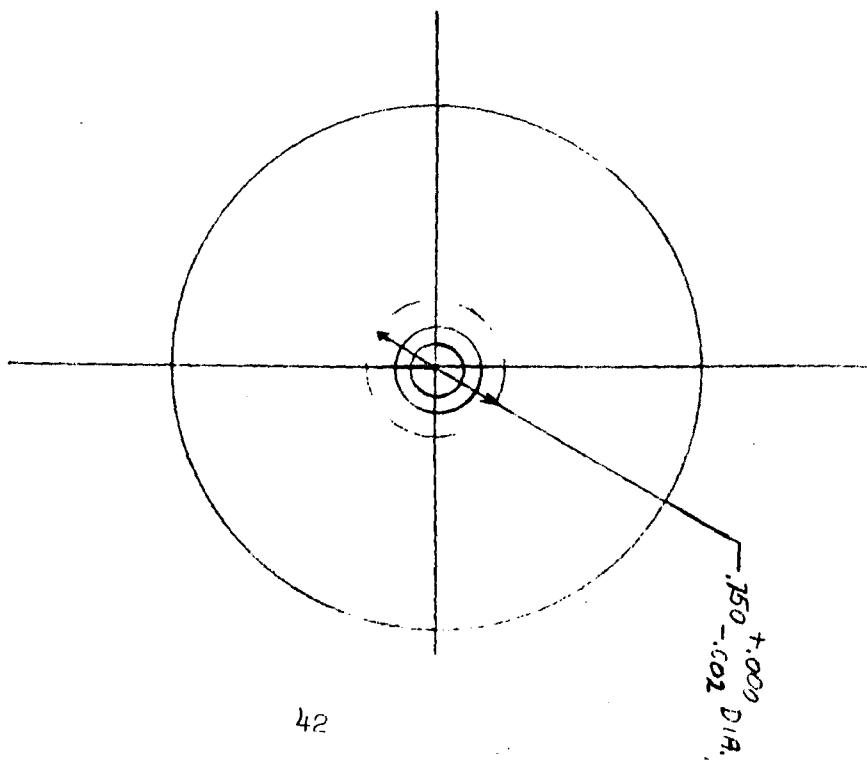
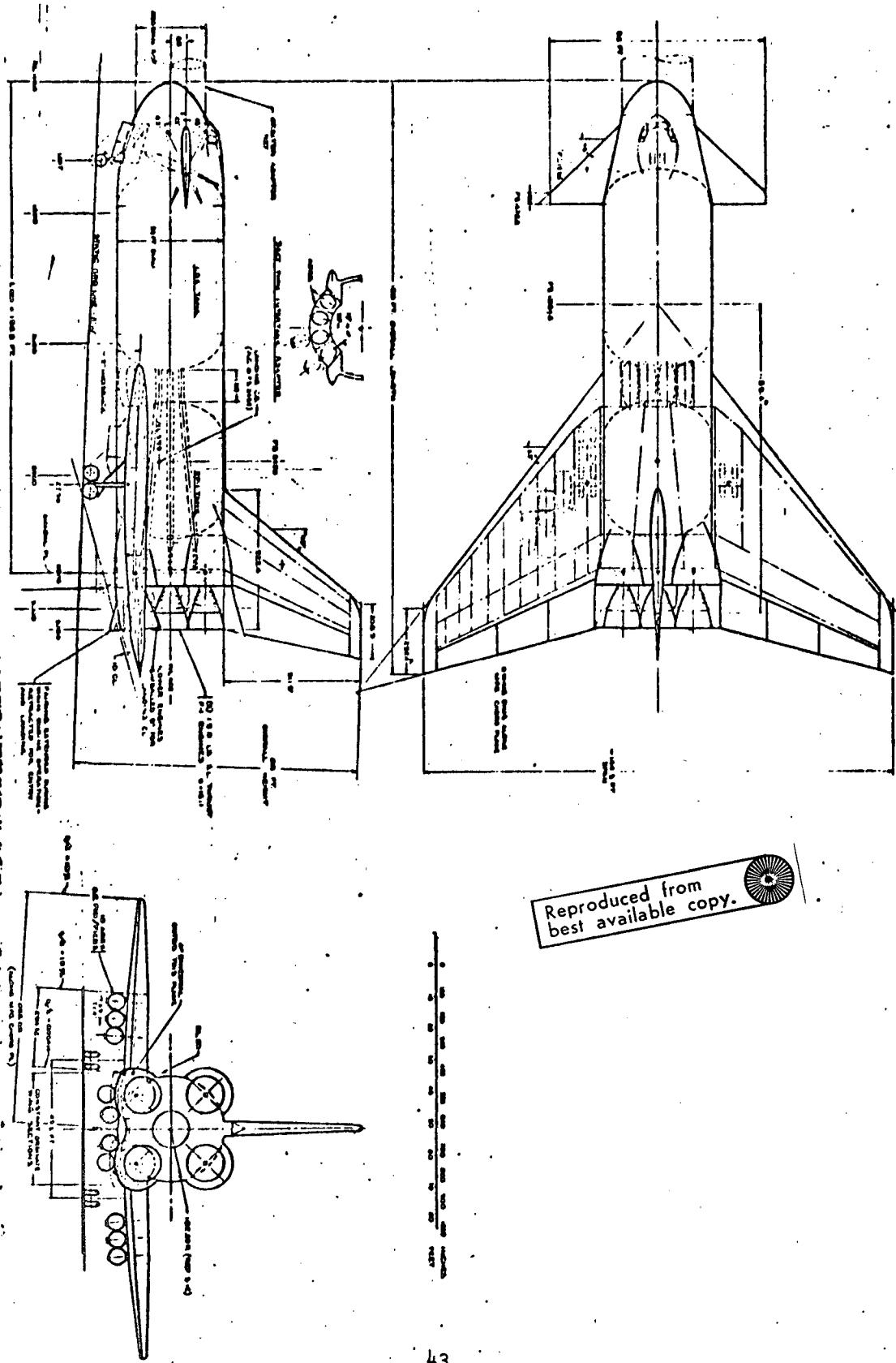


FIGURE 12. PREDESIGN OF THE B-1B/E3 BOOSTER



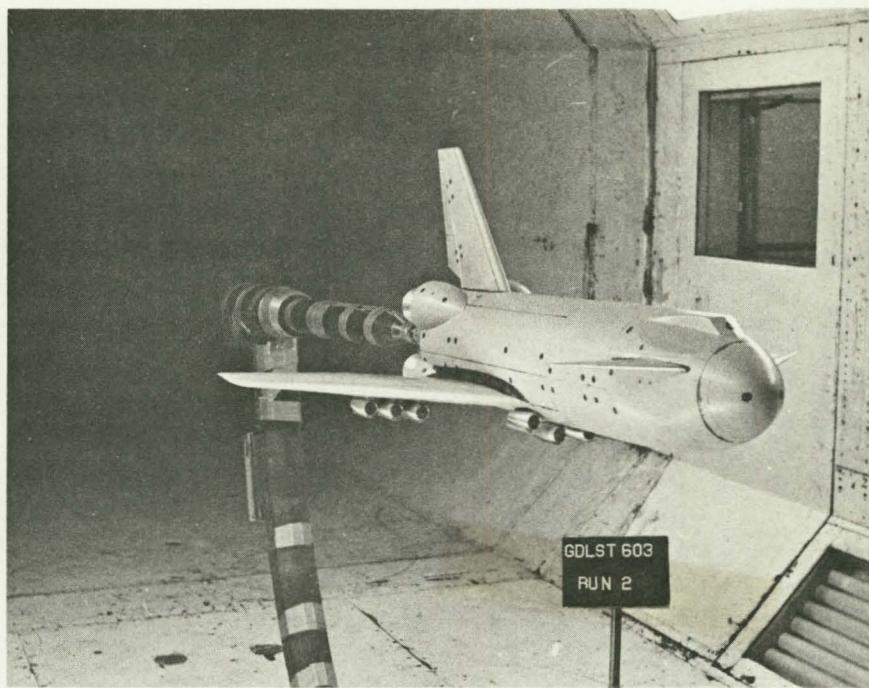


Figure 13. B-18E3 Configuration  $B_{30}^W_{23}C_{10}V_{14}E_{38}$

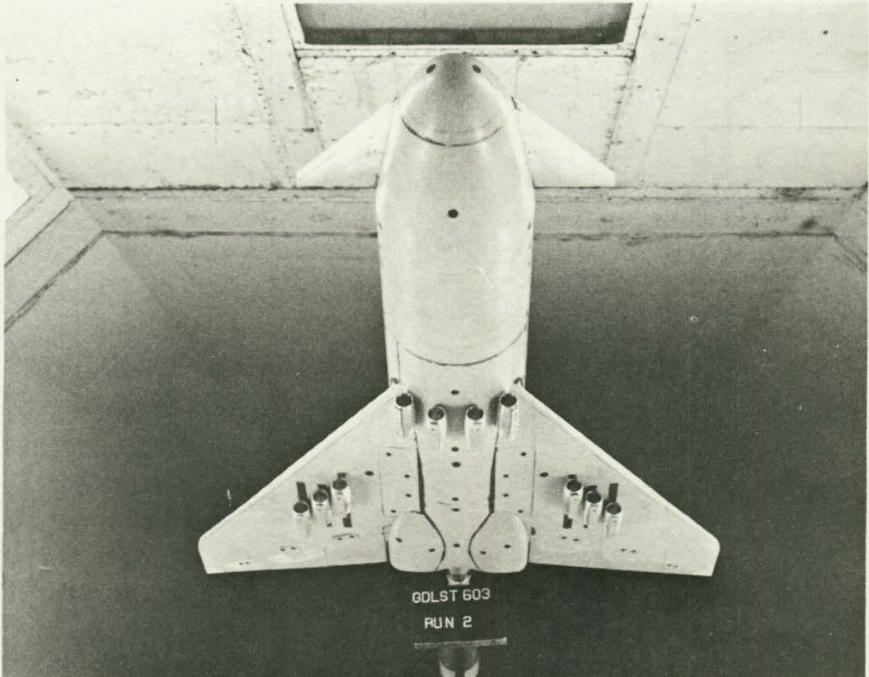


Figure 14. B-18E3 Showing Lower Surface

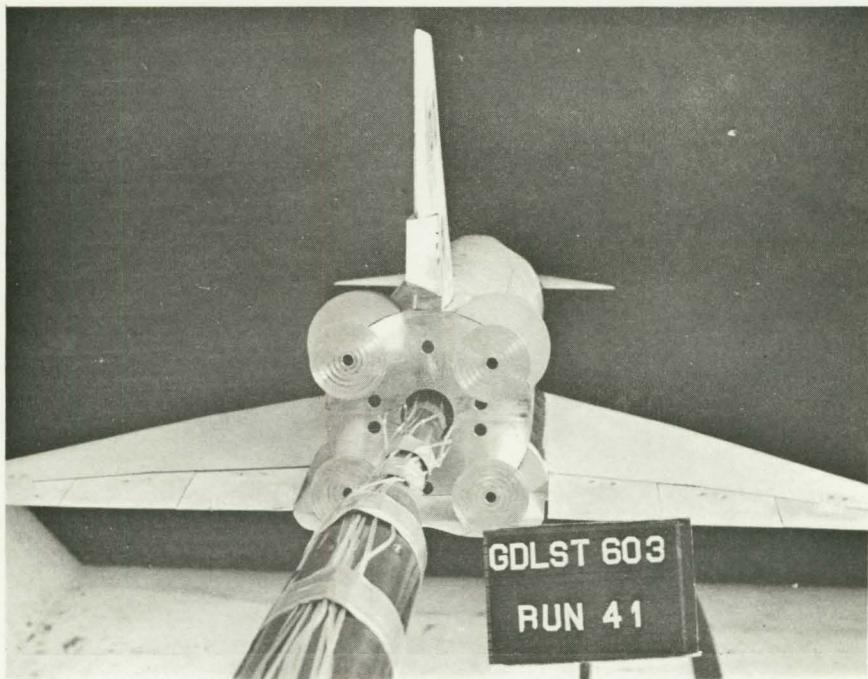


Figure 15. Lower Rudder Deflection and Close Up of Base Region.

Reproduced from  
best available copy.

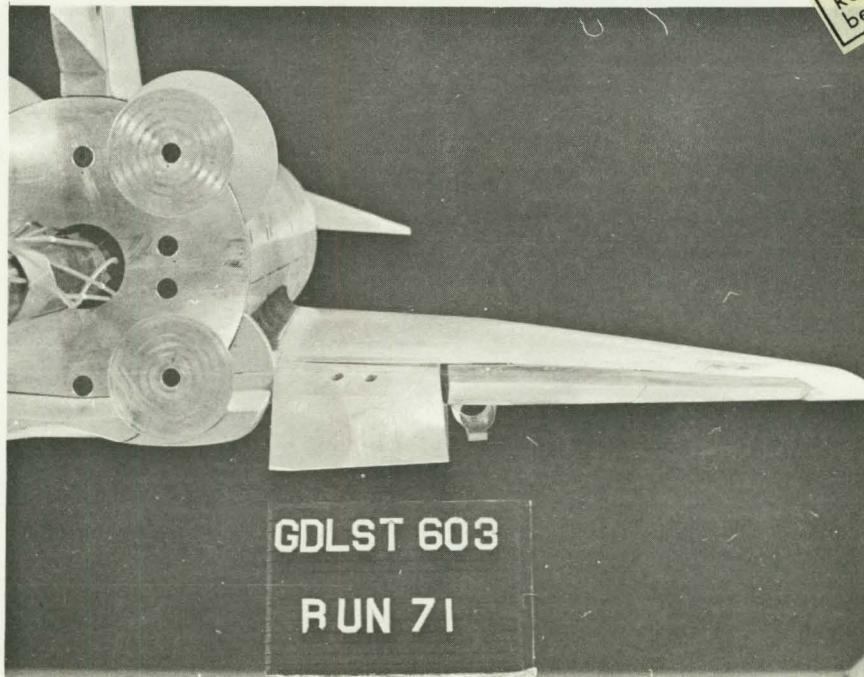


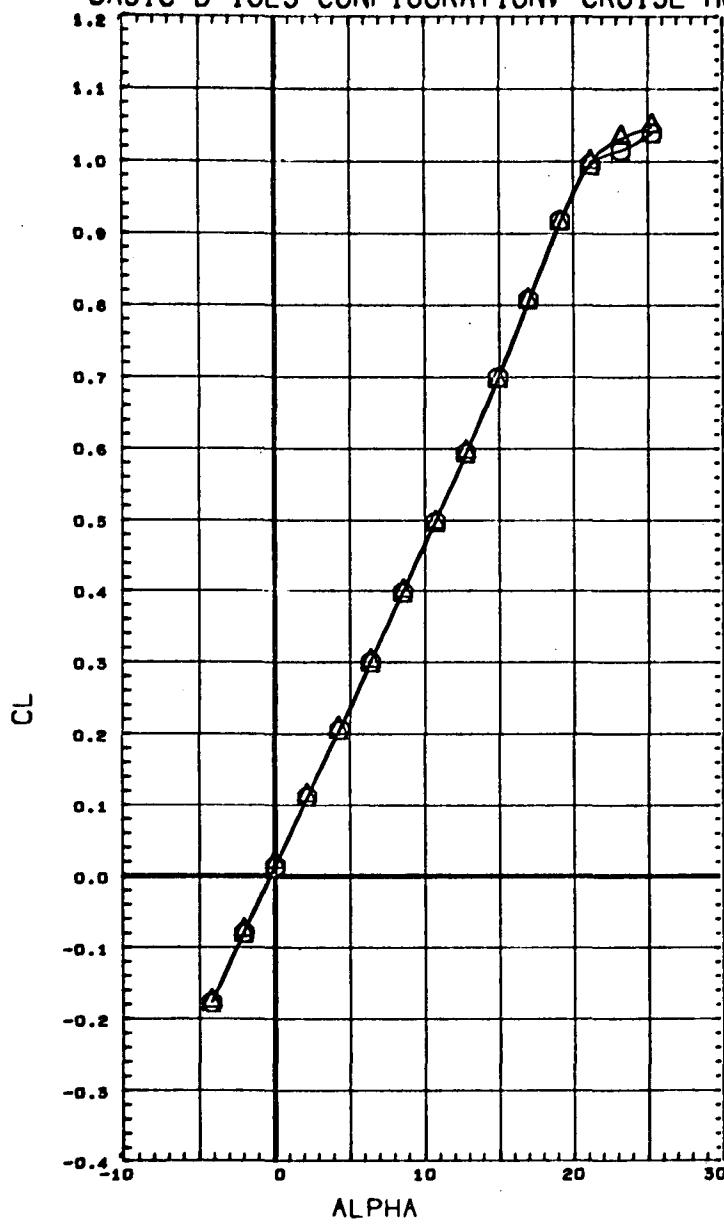
Figure 16. Split Elevon Deflections.

P L O T T E D      D A T A

---

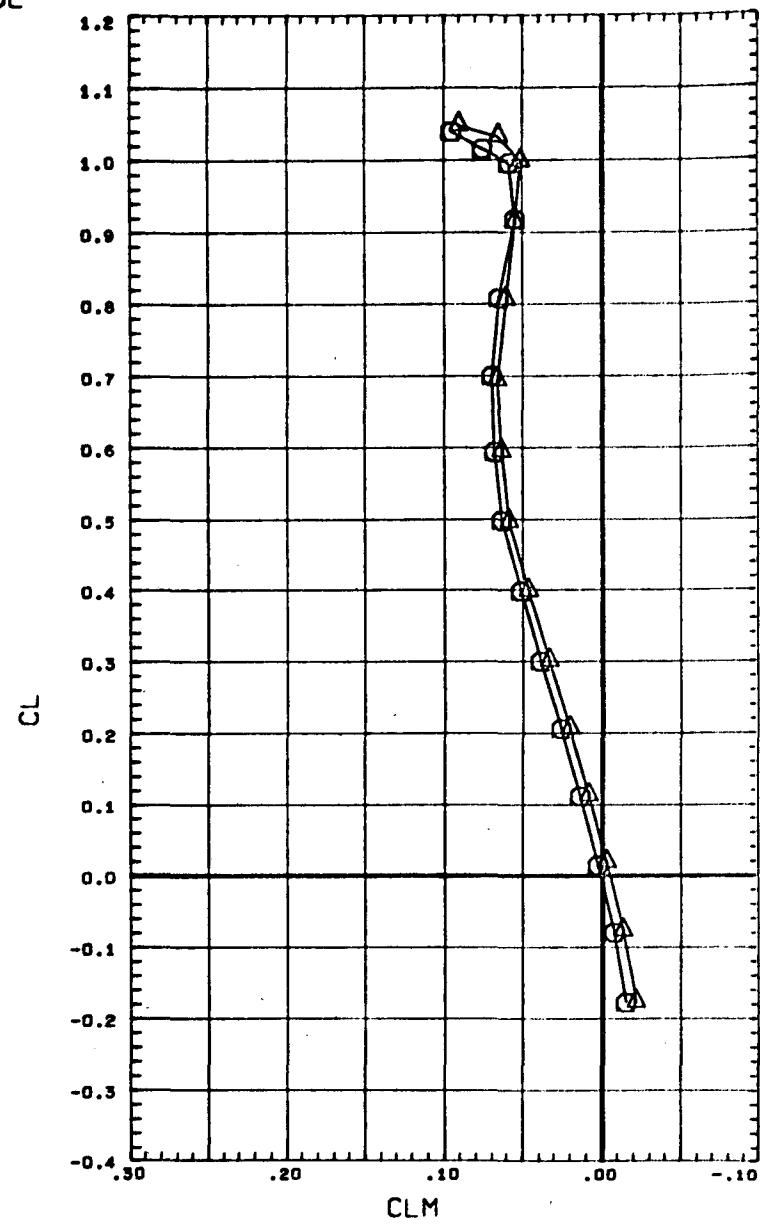
Tabulations of the plotted data and corresponding source data are available from SADSAC Operations.

BASIC B-18E3 CONFIGURATION, CRUISE MODE



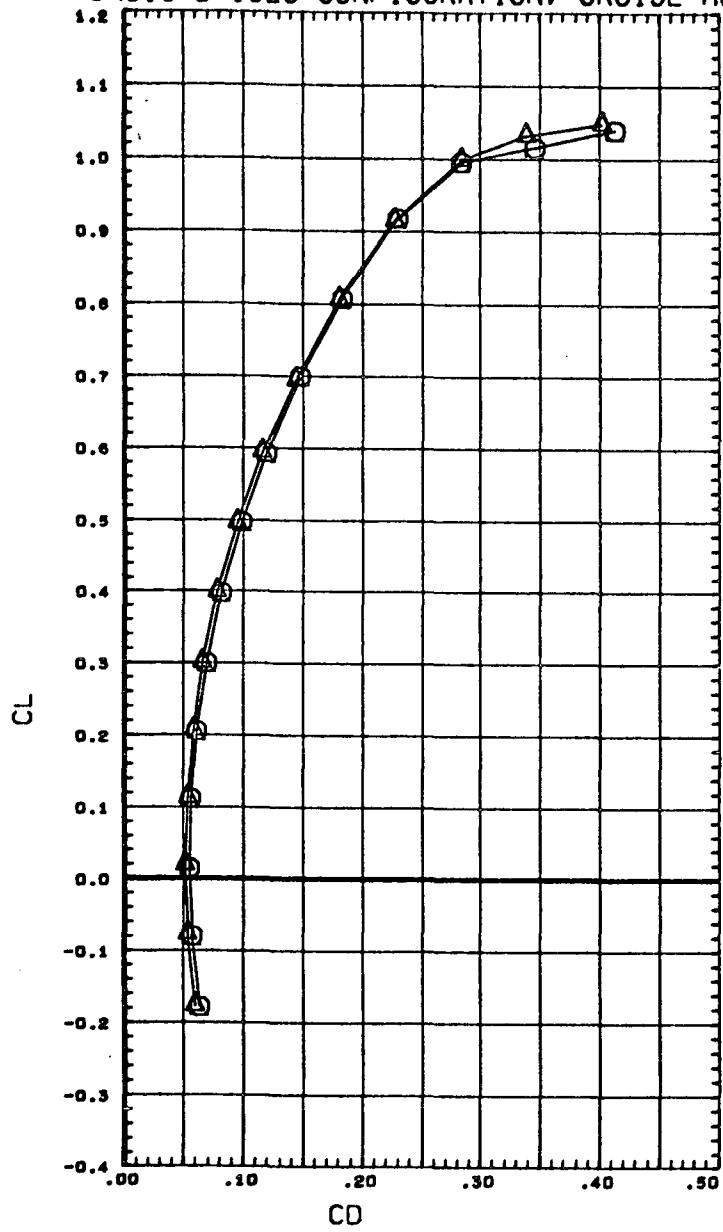
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 (AD6003) Q GDLST 603-0 B30W23C10V14E38

MACH 0.201



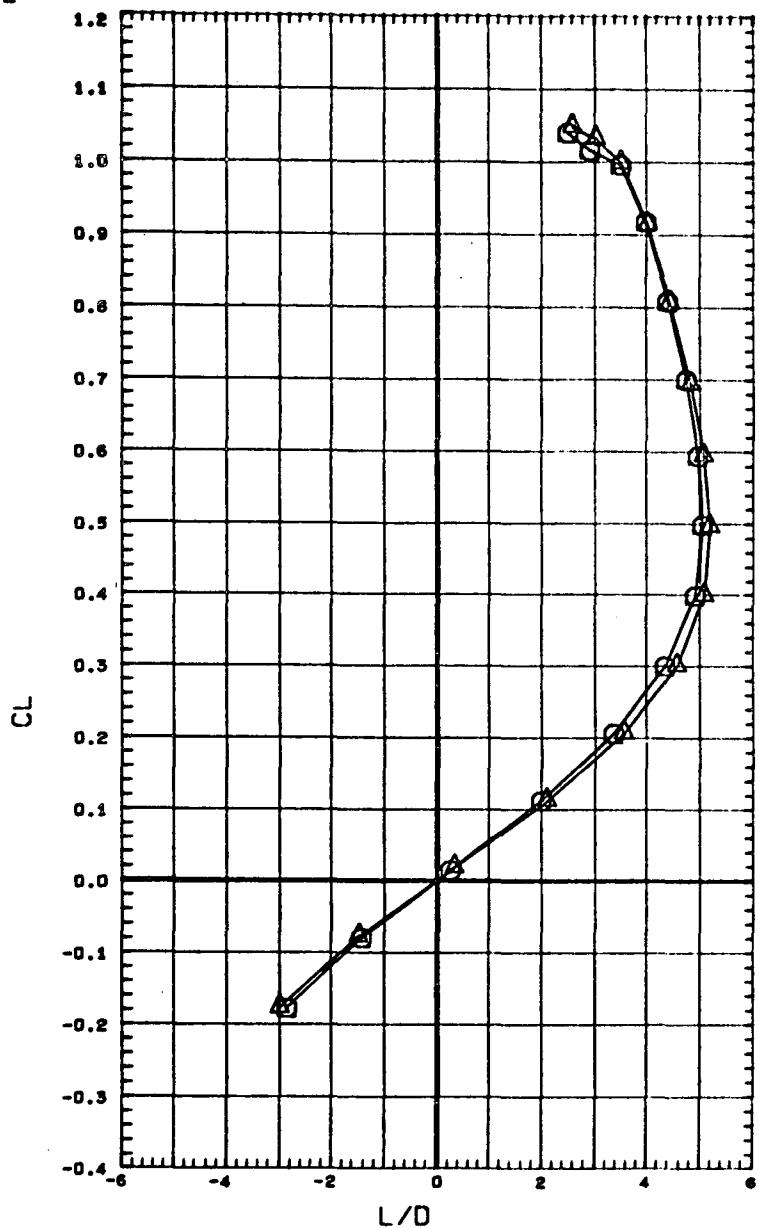
BETA ELEVTR CANARD AILRON REFERENCE INFORMATION  
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 5.000 0.000 0.000 0.000 LREF 16.1880 IN.  
 BREF 34.6320 IN.  
 XMRP 29.0780 IN.  
 YMRP 0.0000 IN.  
 ZMRP 0.0000 IN.  
 SCALE 0.0200

# BASIC B-18E3 CONFIGURATION, CRUISE MODE



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 (AD6003) A GDLST 603-D B30W23C10V14E38

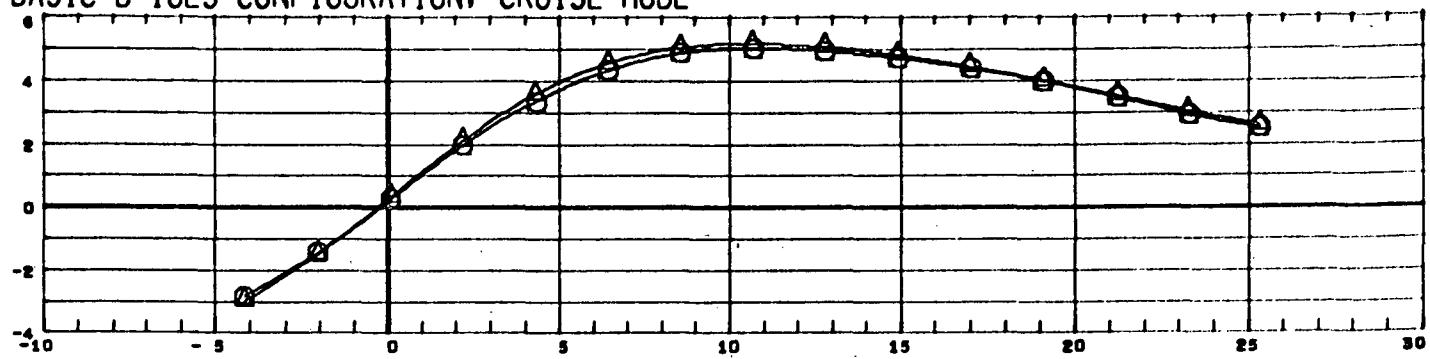
MACH 0.201



BETA	ELEVTR	CANARD	AILRON	REFERENCE INFORMATION
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				BREF 34.6320 IN.
				XMRP 29.0780 IN.
				YMRP 0.0000 IN.
				ZMRP 0.0000 IN.
				SCALE 0.0200

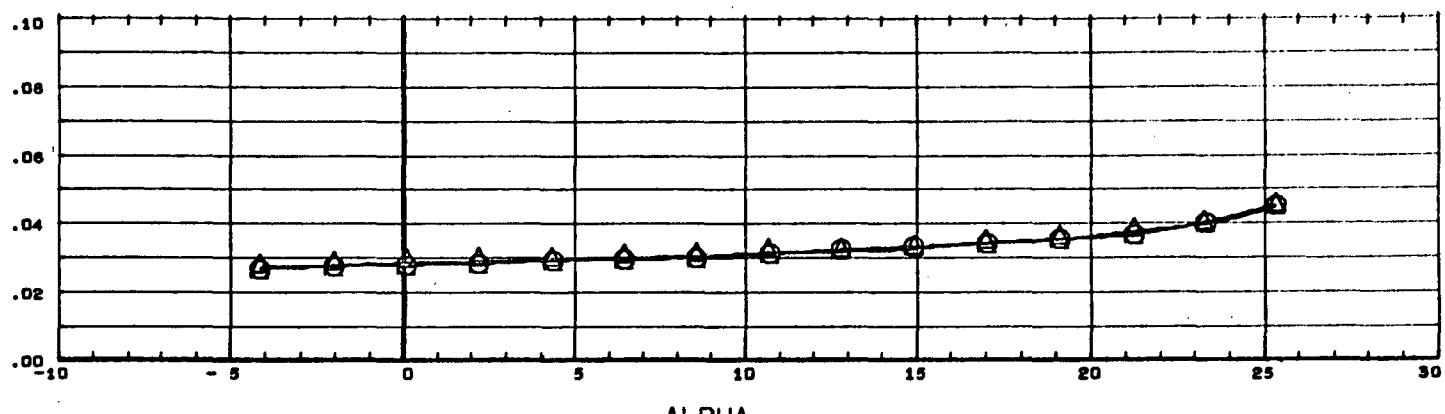
BASIC B-18E3 CONFIGURATION, CRUISE MODE

L/D



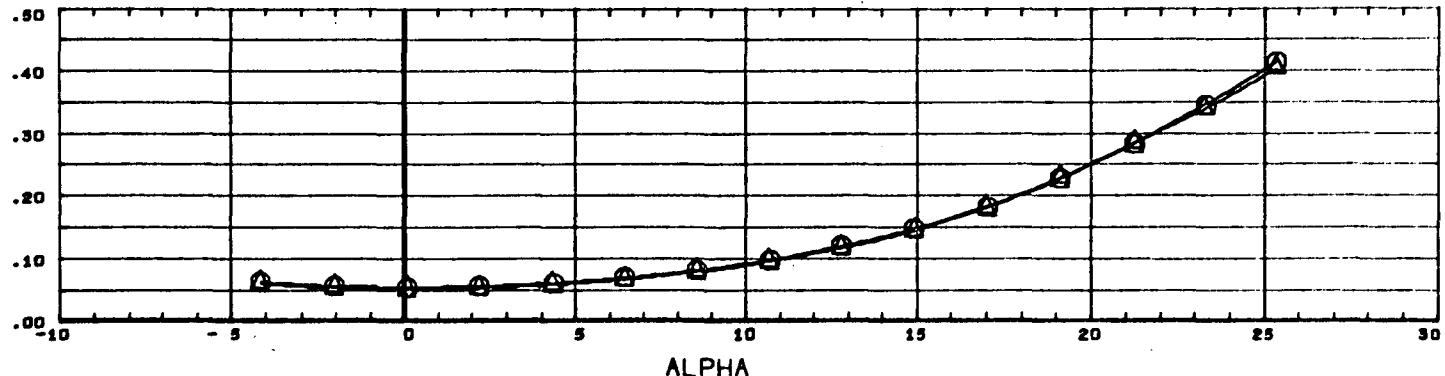
ALPHA

CAB



ALPHA

CD



ALPHA

DATA SET SYMBOL    CONFIGURATION DESCRIPTION  
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 (AD6003)      GDLST 603-0    B30W23C10V14E38

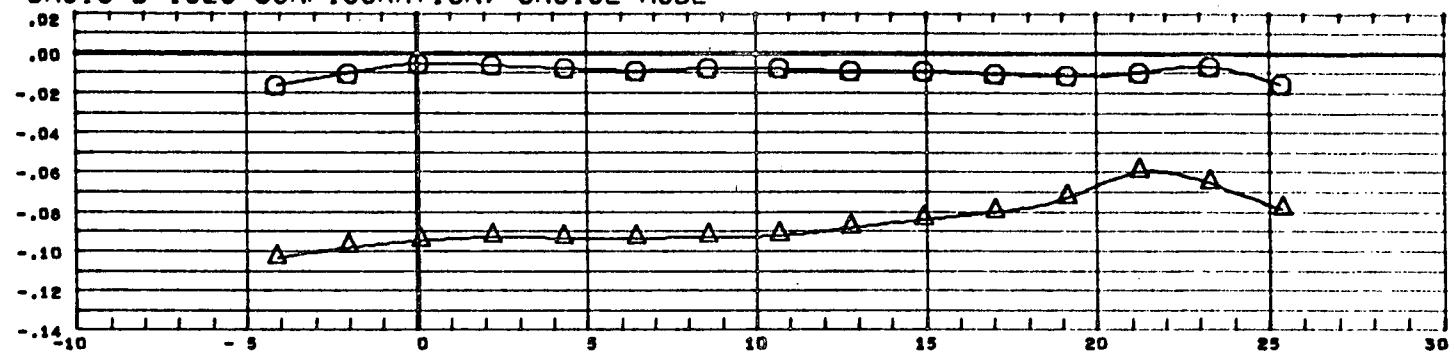
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				BREF	34.6320
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				YMRP	0.0000
				ZMRP	0.0000
				SCALE	0.0200

MACH      0.201

PAGE      3

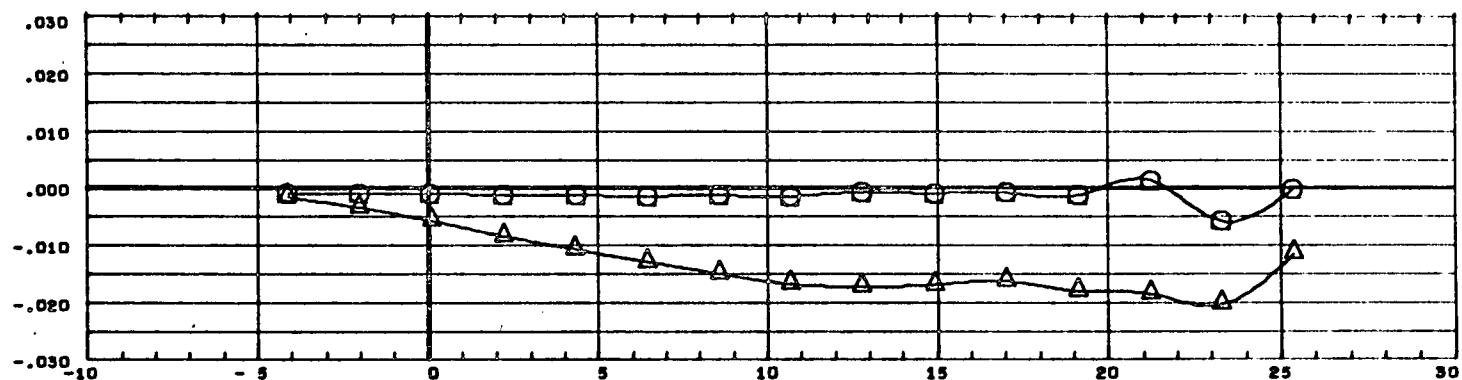
# BASIC B-18E3 CONFIGURATION, CRUISE MODE

CY



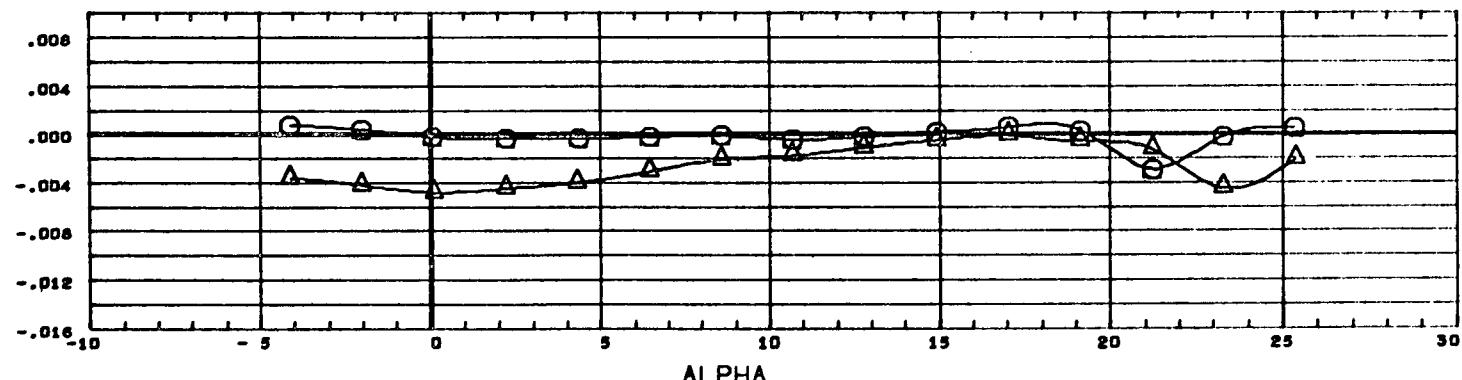
ALPHA

CSL



ALPHA

CLN



ALPHA

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 (AD6003) GDLST 603-0 B30W23C10V14E38

BETA	ELEVTR	RUDDER	AILRON	REFERENCE	INFORMATION
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5.000	0.000	0.000	0.000	LREF	16.1880 IN.
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				YHRP	0.0000 IN.
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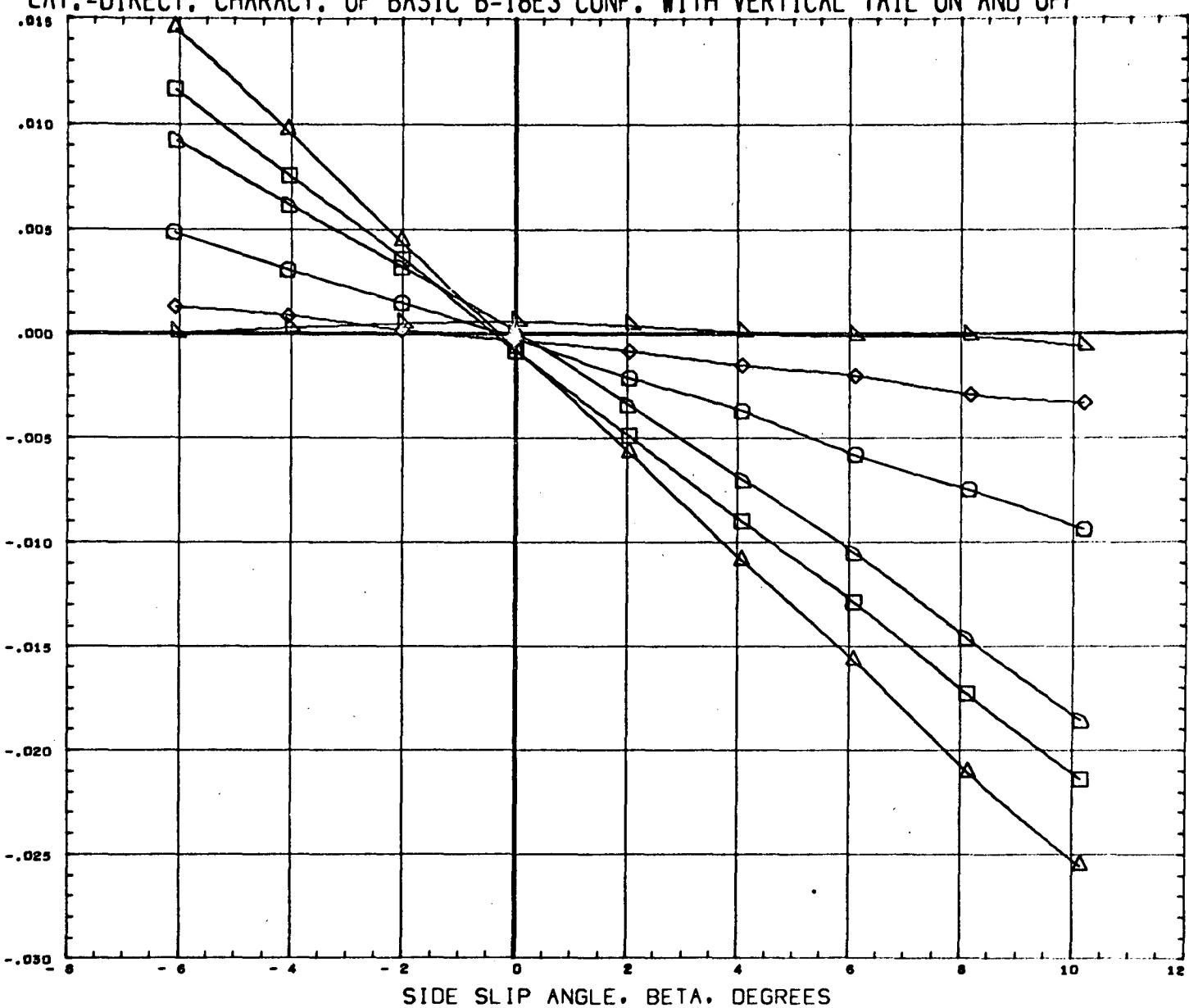
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LAT.-DIRECT. CHARACT. OF BASIC B-18E3 CONF. WITH VERTICAL TAIL ON AND OFF

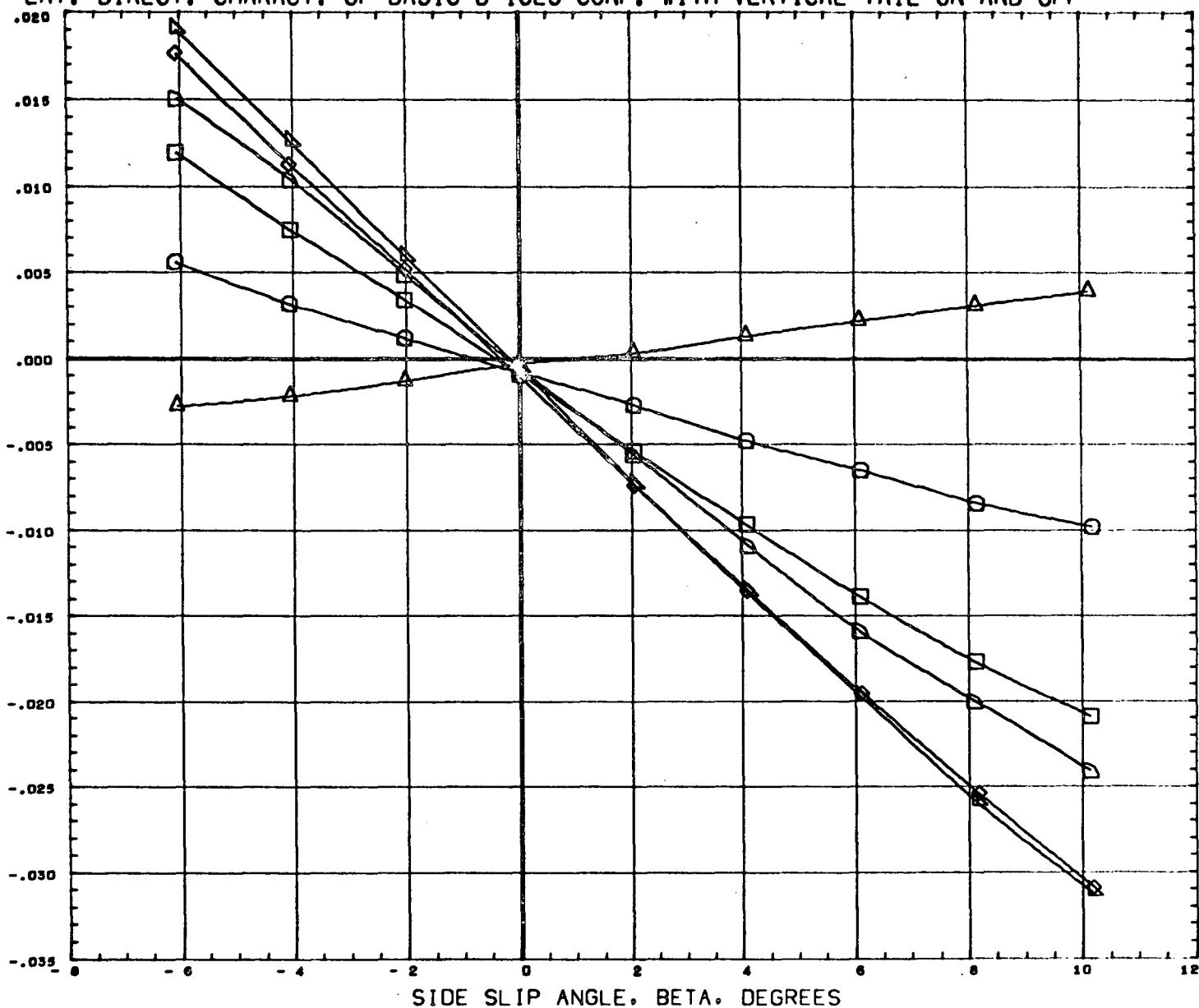
YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



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(RD6056)	GDLST 603-0 B30W23C10E38	0.000	0.000	0.000	0.000	LREF 16.1880 IN.
(RD6005)	GDLST 603-0 B30W23C10V14E38	10.000	0.000	0.000	0.000	BREF 34.6320 IN.
(RD6057)	GDLST 603-0 B30W23C10E38	10.000	0.000	0.000	0.000	XMRP 29.0780 IN.
(RD6006)	GDLST 603-0 B30W23C10V14E38	15.000	0.000	0.000	0.000	YMRP 0.0000 IN.
(RD6058)	GDLST 603-0 B30W23C10E38	15.000	0.000	0.000	0.000	ZMRP 0.0000 IN.
MACH	0.201					SCALE 0.0200

LAT.-DIRECT. CHARACT. OF BASIC B-18E3 CONF. WITH VERTICAL TAIL ON AND OFF

ROLLING MOMENT COEFFICIENT. CSL (STABILITY AXIS)



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	ELEVTR	RUDDER	AIRLON	REFERENCE	INFORMATION
(RD6004)	GOLST 603-0 B30W23C10V14E38	0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN.
(RD6056)	GOLST 603-0 B30W23C10E38	0.000	0.000	0.000	0.000	LREF	16.1880 IN.
(RD6005)	GOLST 603-0 B30W23C10V14E38	10.000	0.000	0.000	0.000	BREF	34.6320 IN.
(RD6057)	GOLST 603-0 B30W23C10E38	10.000	0.000	0.000	0.000	XMRP	29.0780 IN.
(RD6006)	GOLST 603-0 B30W23C10V14E38	15.000	0.000	0.000	0.000	YMRP	0.0000 IN.
(RD6058)	GOLST 603-0 B30W23C10E38	15.000	0.000	0.000	0.000	ZMRP	0.0000 IN.
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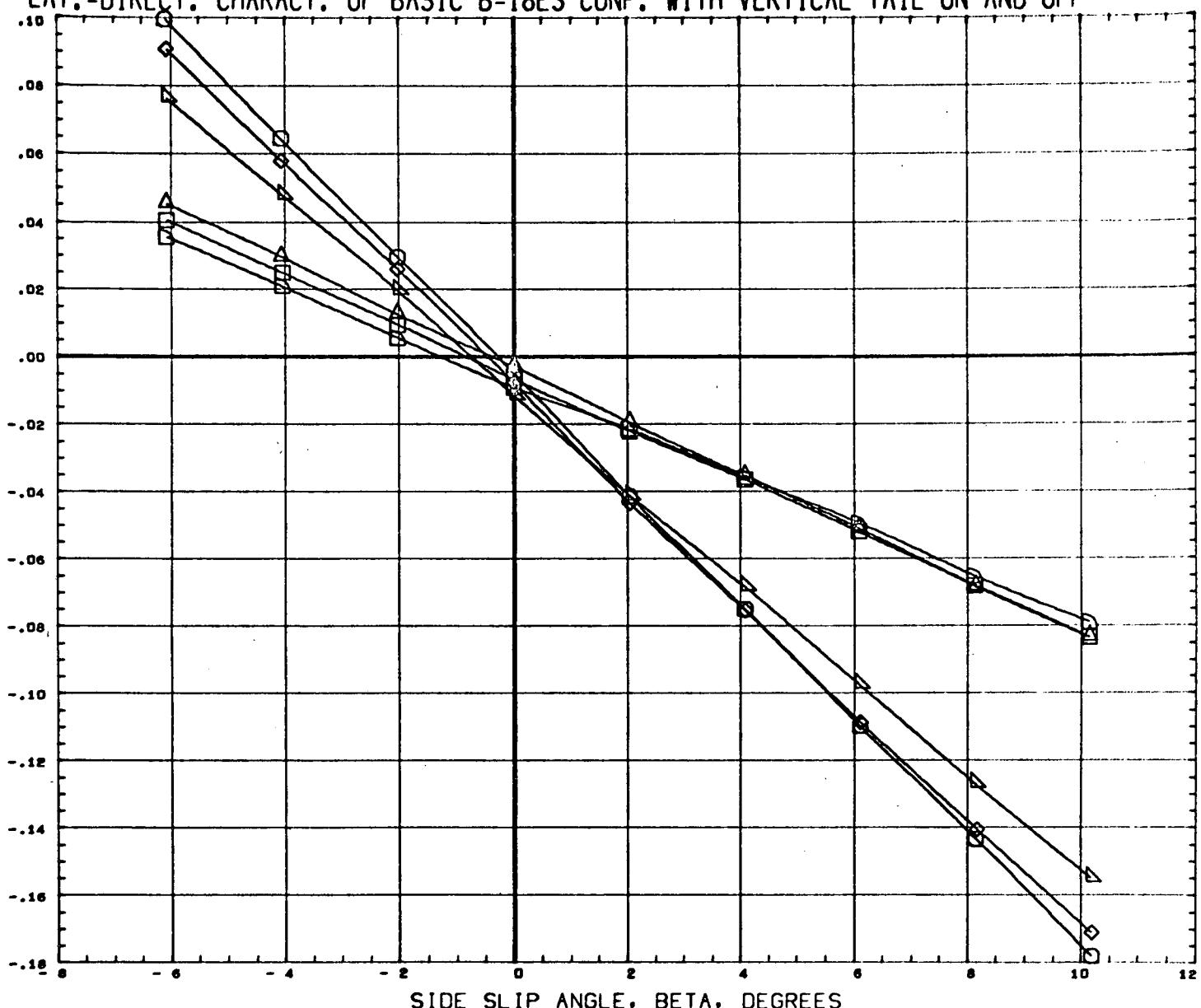
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6

LAT.-DIRECT. CHARACT. OF BASIC B-18E3 CONF. WITH VERTICAL TAIL ON AND OFF

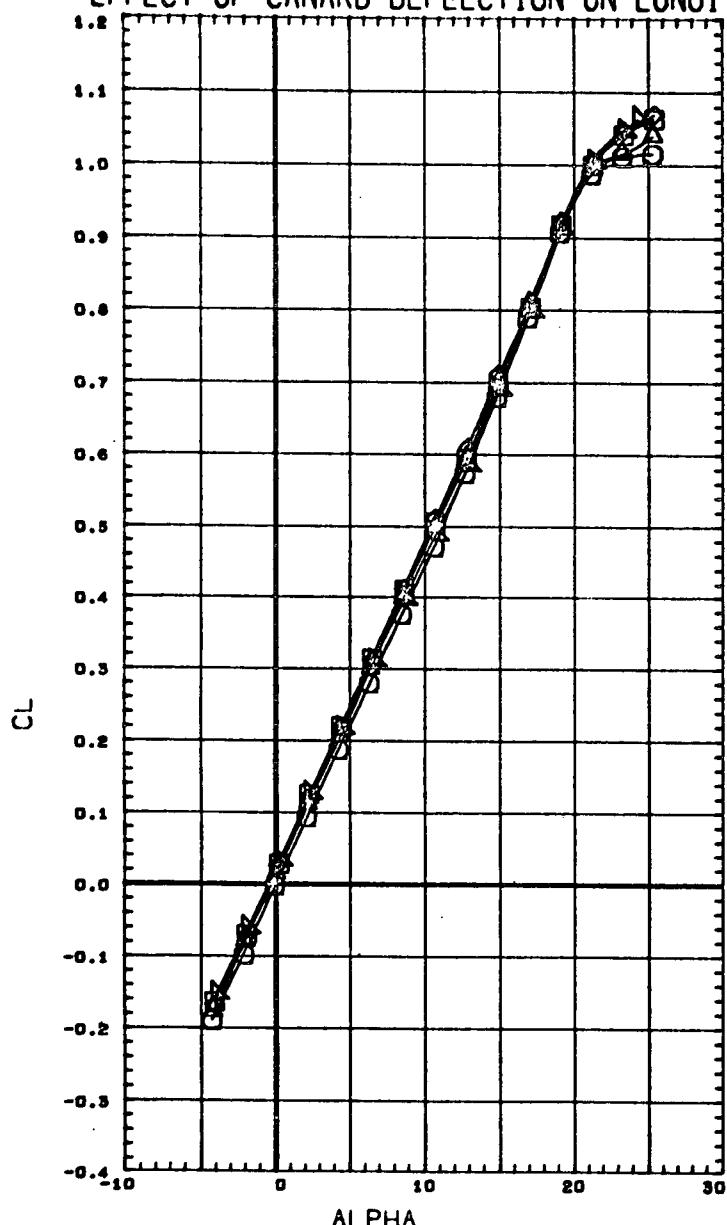
LATERAL FORCE COEFFICIENT, CY



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	ELEVTR	RUDDER	AILRON	REFERENCE INFORMATION
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(RD6056)	GDLST 603-0 B30W23C10E38	0.000	0.000	0.000	0.000	LREF 16.1880 IN.
(RD6005)	GDLST 603-0 B30W23C10V14E38	10.000	0.000	0.000	0.000	BREF 34.6320 IN.
(RD6057)	GDLST 603-0 B30W23C10E38	10.000	0.000	0.000	0.000	XMRP 29.0780 IN.
(RD6006)	GDLST 603-0 B30W23C10V14E38	15.000	0.000	0.000	0.000	YMRP 0.0000 IN.
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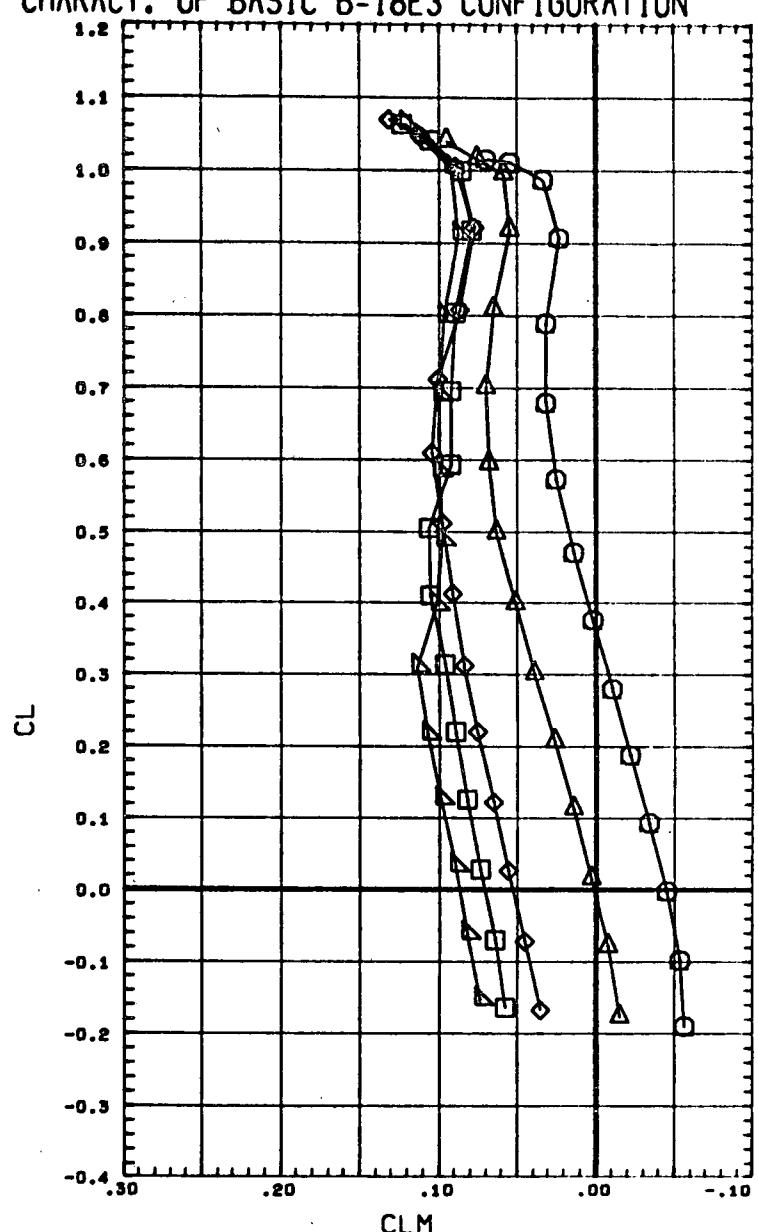
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# EFFECT OF CANARD DEFLECTION ON LONGIT. CHARACT. OF BASIC B-18E3 CONFIGURATION



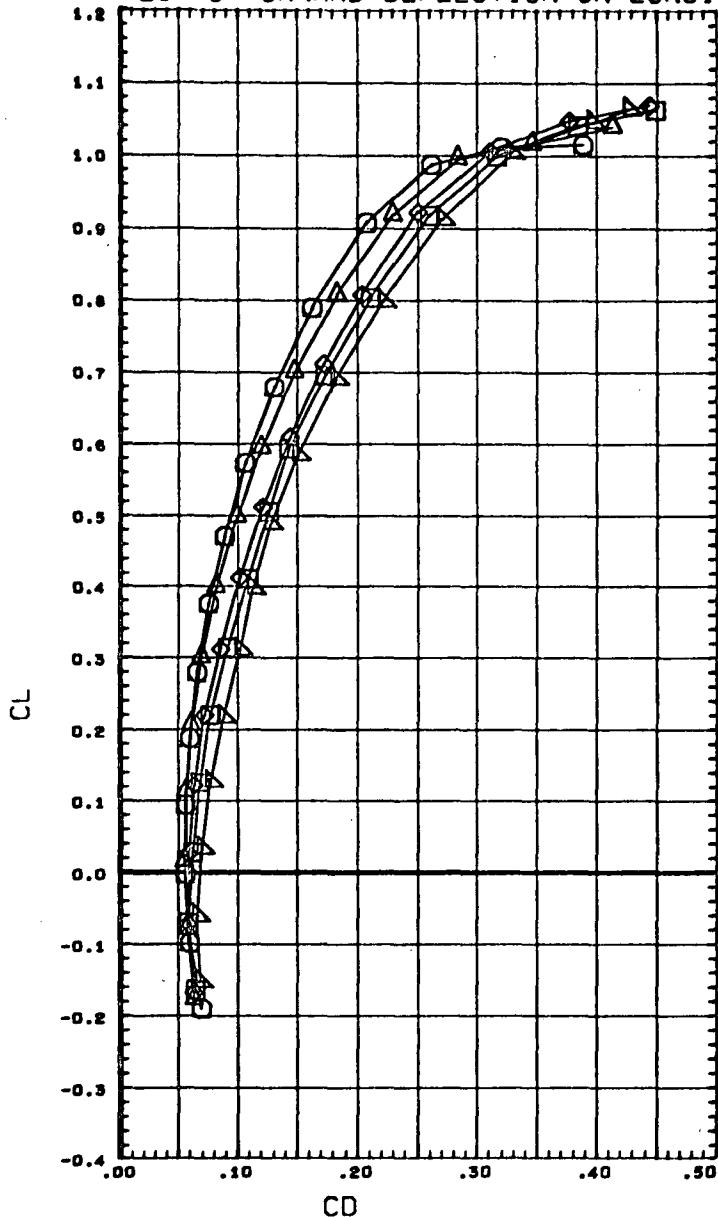
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 (AD6007) GDLST 603-0 B30W23C10V14E38  
 (AD6011) GDLST 603-0 B30W23C10V14E38  
 (AD6013) GDLST 603-0 B30W23C10V14E38

MACH 0.201



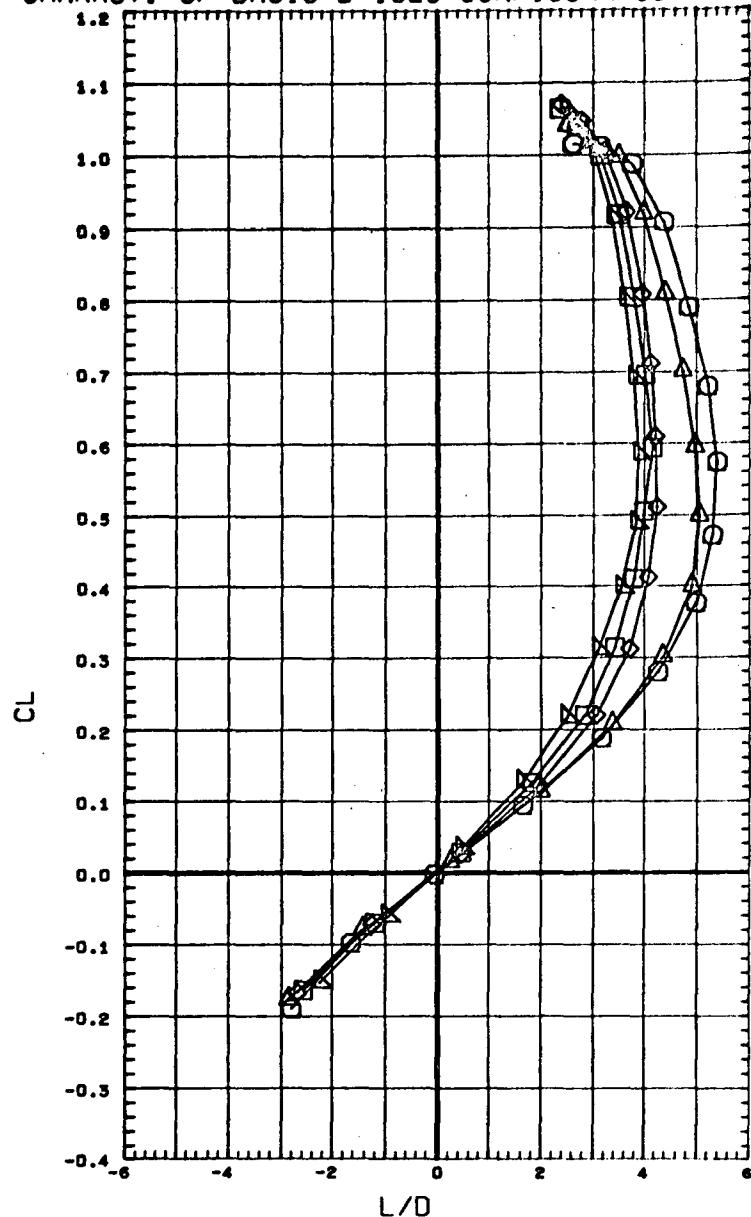
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0.000	0.000	0.000	0.000	LREF	16.1880 IN.
0.000	0.000	10.000	0.000	BREF	34.6320 IN.
0.000	0.000	15.000	0.000	XMRP	29.0780 IN.
0.000	0.000	20.000	0.000	YMRP	0.0000 IN.
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# EFFECT OF CANARD DEFLECTION ON LONGIT. CHARACT. OF BASIC B-18E3 CONFIGURATION



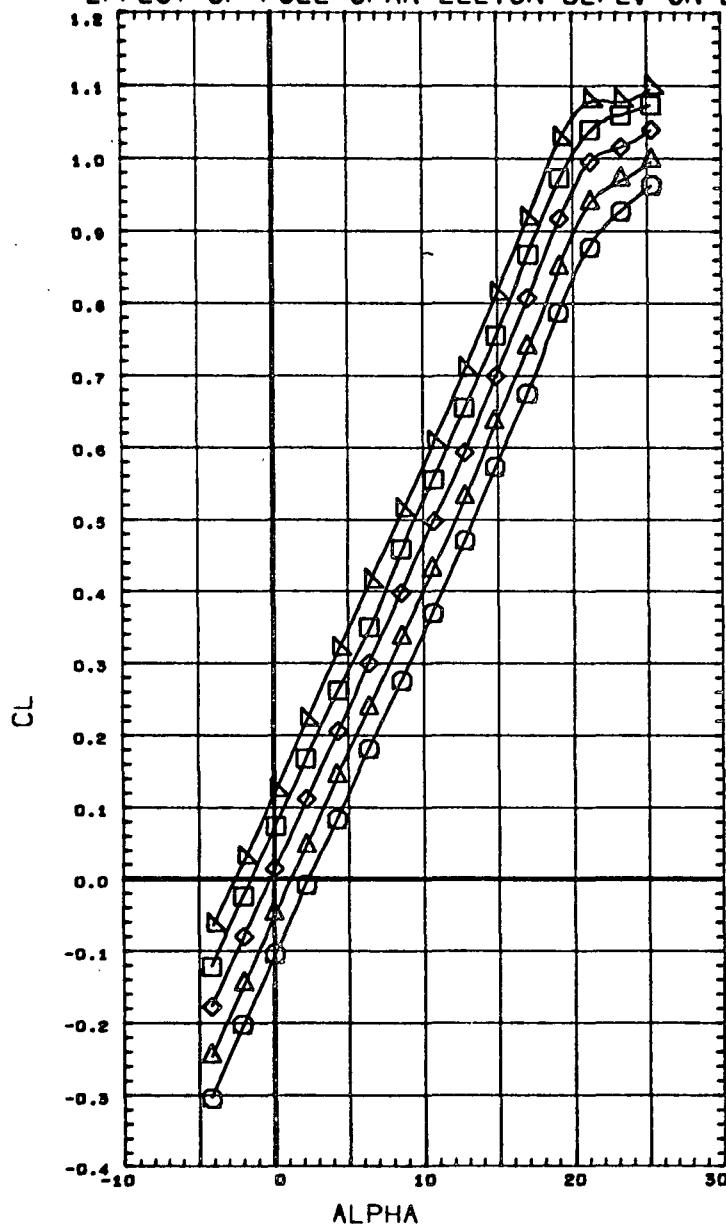
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(AD6007)	GDLST 603-0 B30W23C10V14E38
(AD6011)	GDLST 603-0 B30W23C10V14E38
(AD6013)	GDLST 603-0 B30W23C10V14E38

MACH 0.201



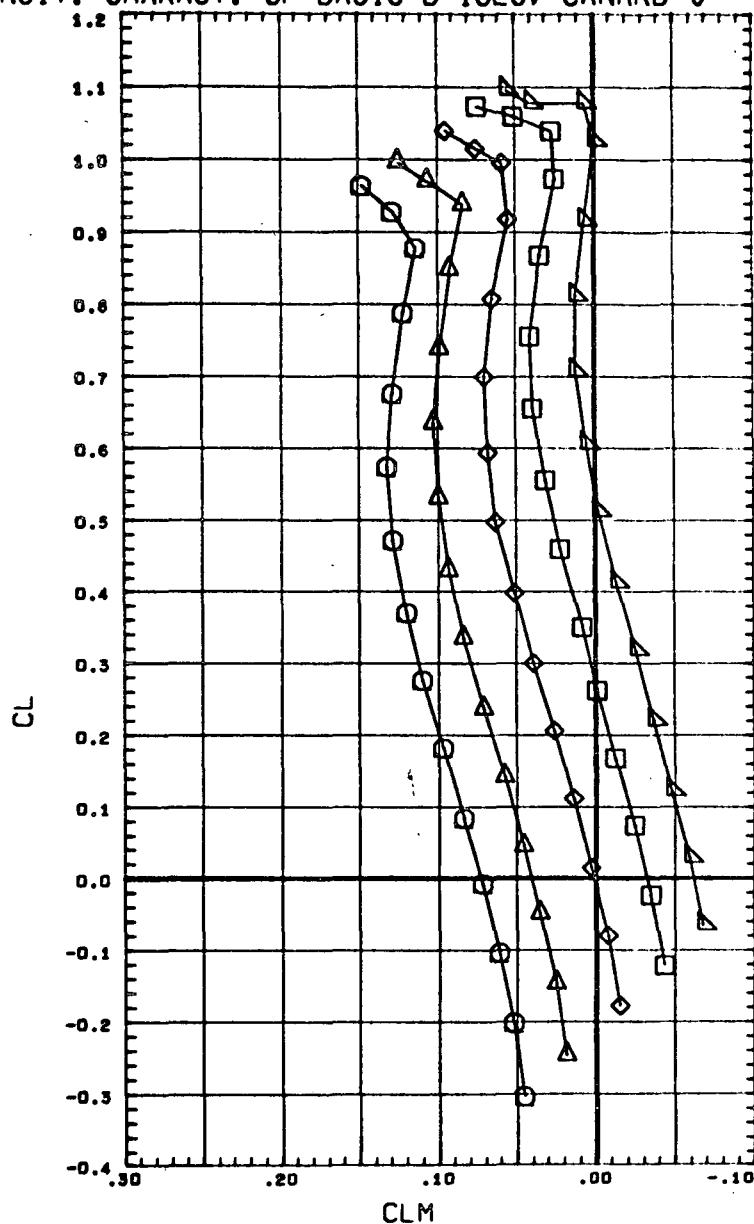
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0.000	0.000	10.000	0.000	BREF	34.6320 IN.
0.000	0.000	15.000	0.000	XMRP	29.0780 IN.
0.000	0.000	20.000	0.000	YMRP	0.0000 IN.
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EFFECT OF FULL SPAN ELEVON DEF'L. ON LONGIT. CHARACT. OF BASIC B-18E3, CANARD=0



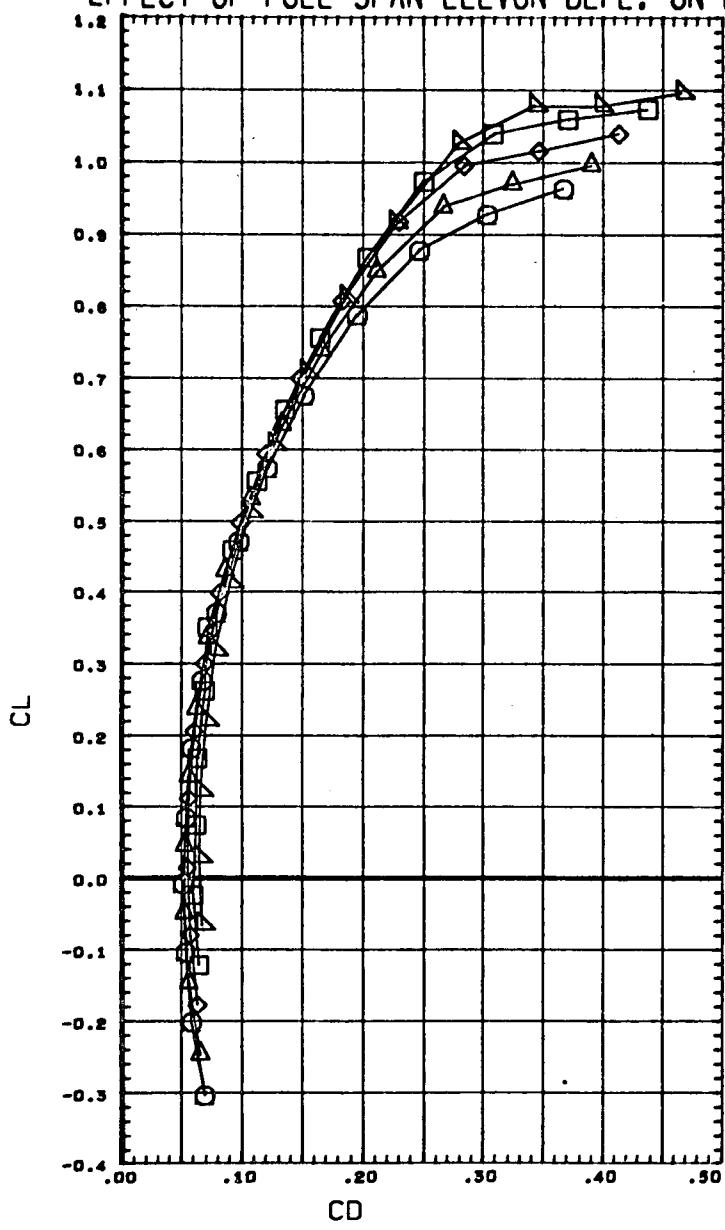
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(AD6002)	GDLST 603-0 B30W23C10V14E38
(AD6024)	GDLST 603-0 B30W23C10V14E38
(AD6026)	GDLST 603-0 B30W23C10V14E38

MACH 0.201



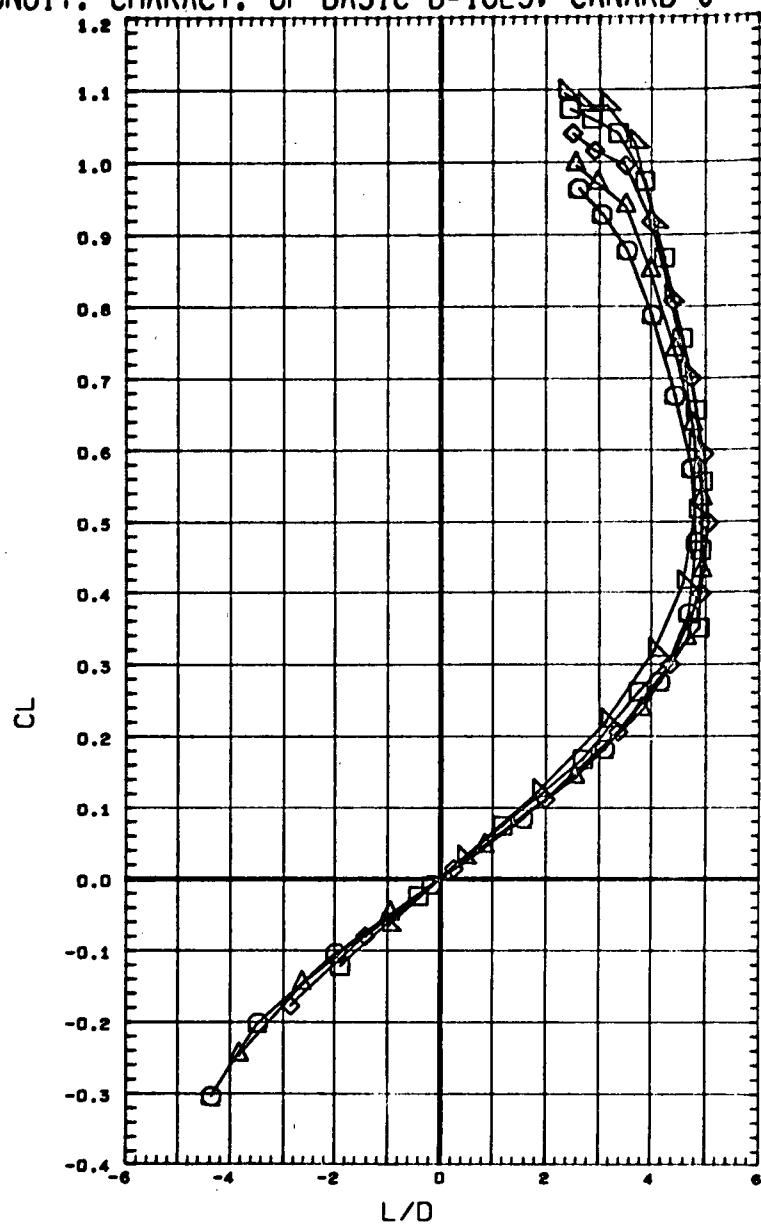
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0.000	5.000	0.000	0.000	XMRP	29.0780 IN.
0.000	10.000	0.000	0.000	YMRP	0.0000 IN.
				ZMRP	0.0000 IN.
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EFFECT OF FULL SPAN ELEVON DEFL. ON LONGIT. CHARACT. OF BASIC B-18E3, CANARD=0



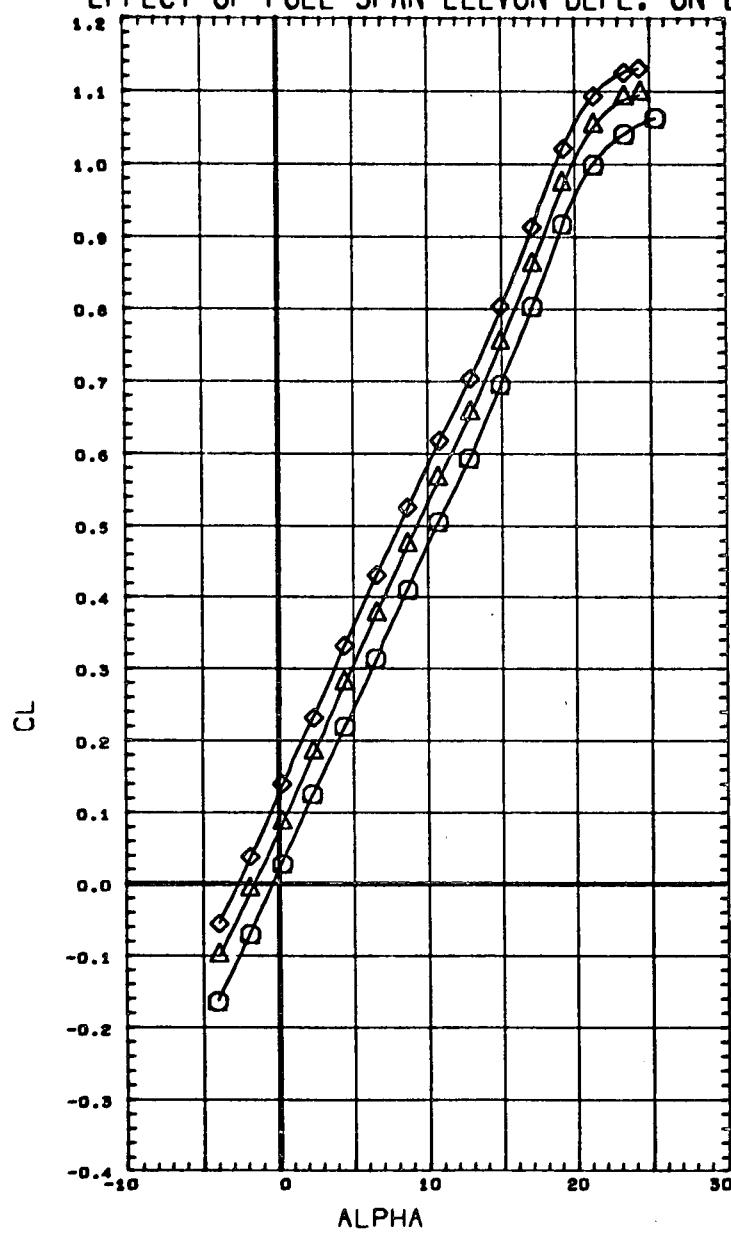
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(AD6002)	◇	GDLST 603-0	B30W23C10V14E38
(AD6024)	△	GDLST 603-0	B30W23C10V14E38
(AD6026)	▽	GDLST 603-0	B30W23C10V14E38

MACH 0.201



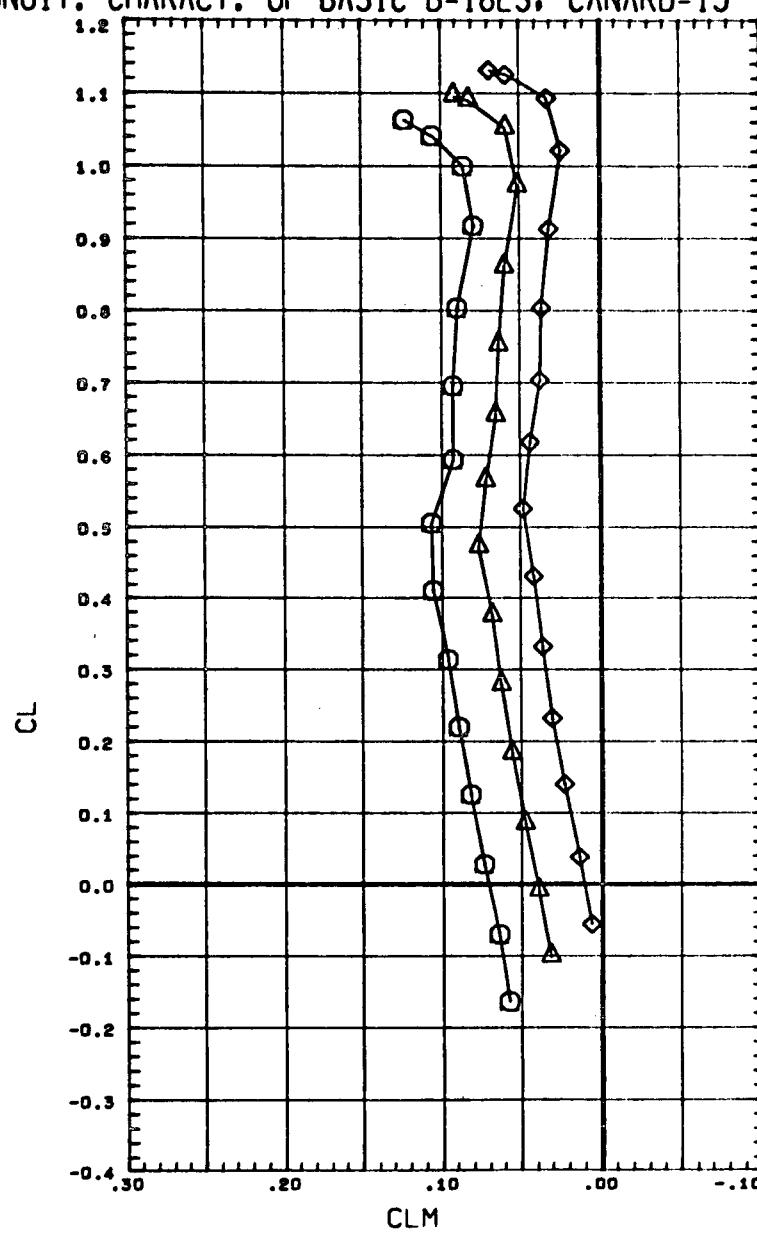
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0.000	5.000	0.000	0.000	XMRP	29.0780
0.000	10.000	0.000	0.000	YMRP	0.0000
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				SCALE	0.0200

EFFECT OF FULL SPAN ELEVON DEF'L. ON LONGIT. CHARACT. OF BASIC B-18E3, CANARD=15



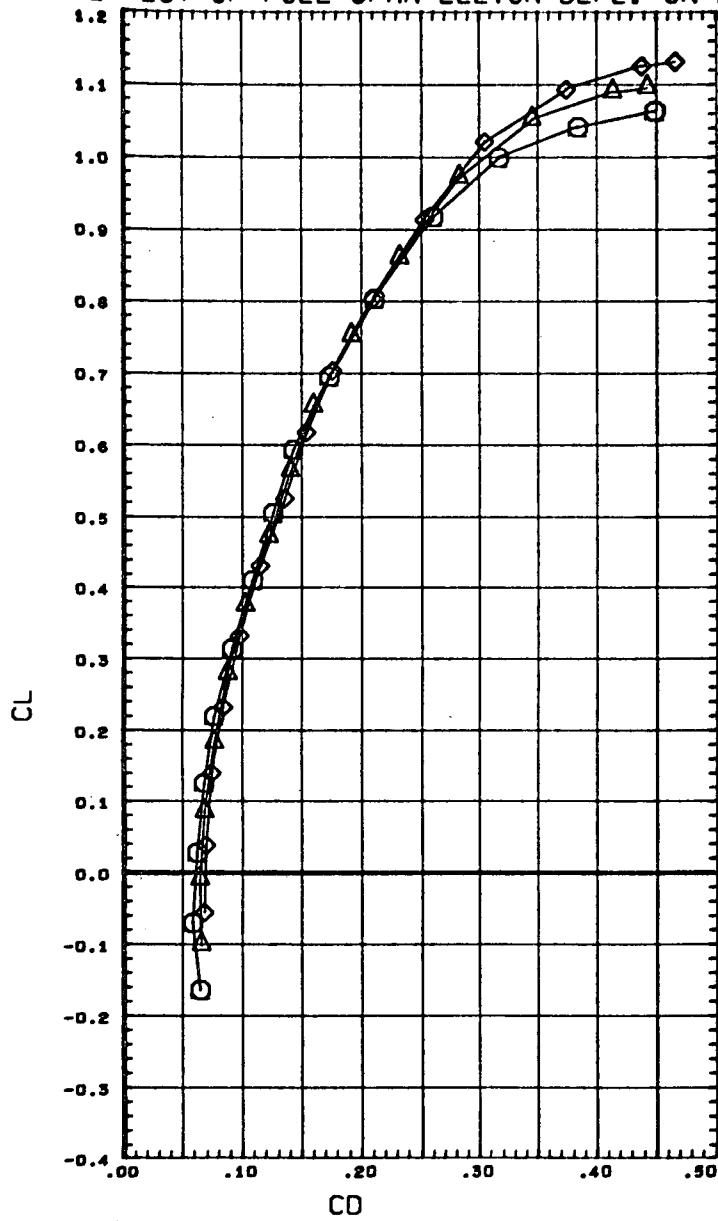
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 (AD6022) A GDLST 603-0 B30W23C10V14E38  
 (AD6020) D GDLST 603-0 B30W23C10V14E38

MACH 0.201



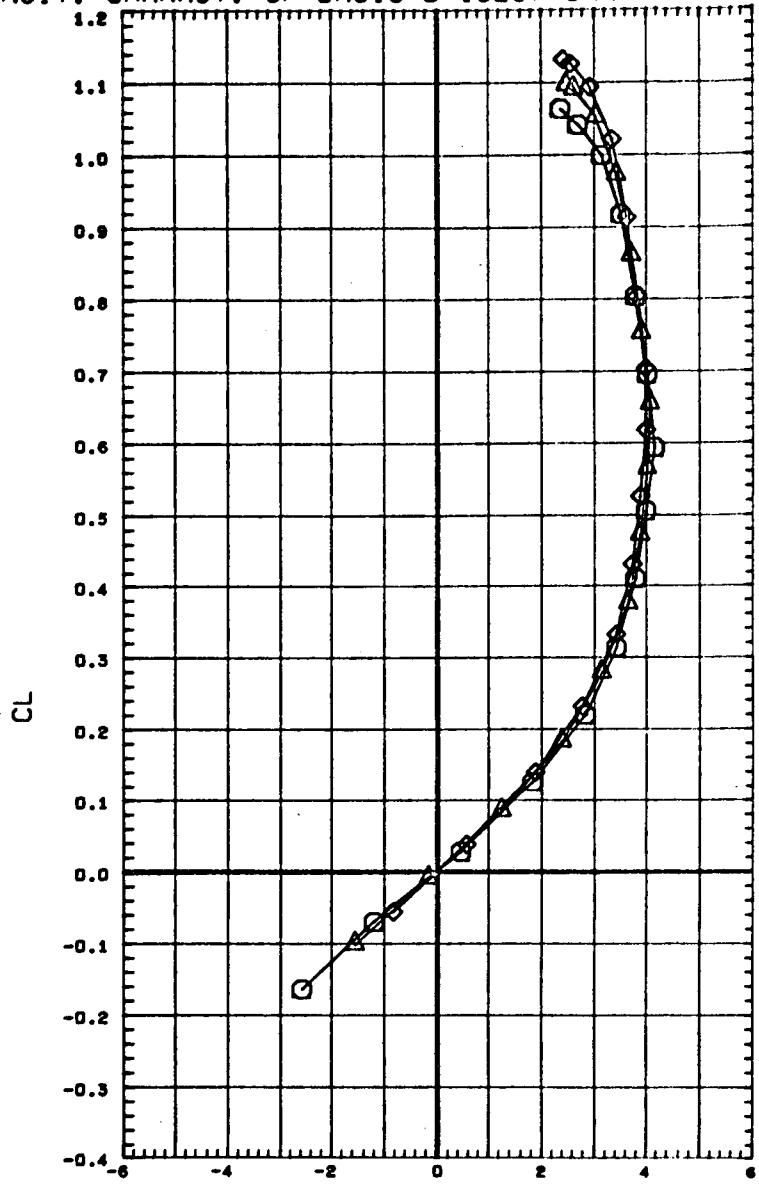
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				YMRP	0.0000 IN.
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EFFECT OF FULL SPAN ELEVON DEFL. ON LONGIT. CHARACT. OF BASIC B-18E3, CANARD=15



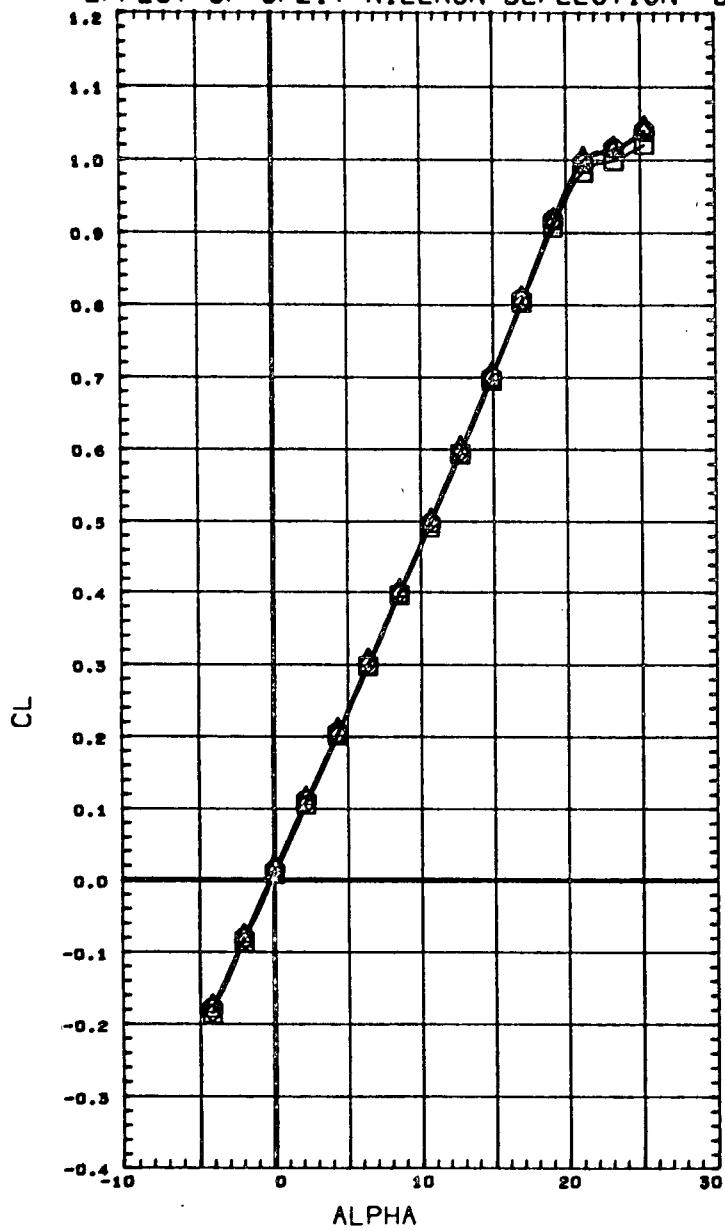
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 (AD6022) GDLST 603-0 B30W23C10V14E38.  
 (AD6020) GDLST 603-0 B30W23C10V14E38

MACH 0.201



BETA	ELEVTR	CANARD	AILRON	REFERENCE INFORMATION
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0.000	5.000	15.000	0.000	LREF 16.1880 IN.
0.000	10.000	15.000	0.000	BREF 34.6320 IN.
				XMRP 29.0780 IN.
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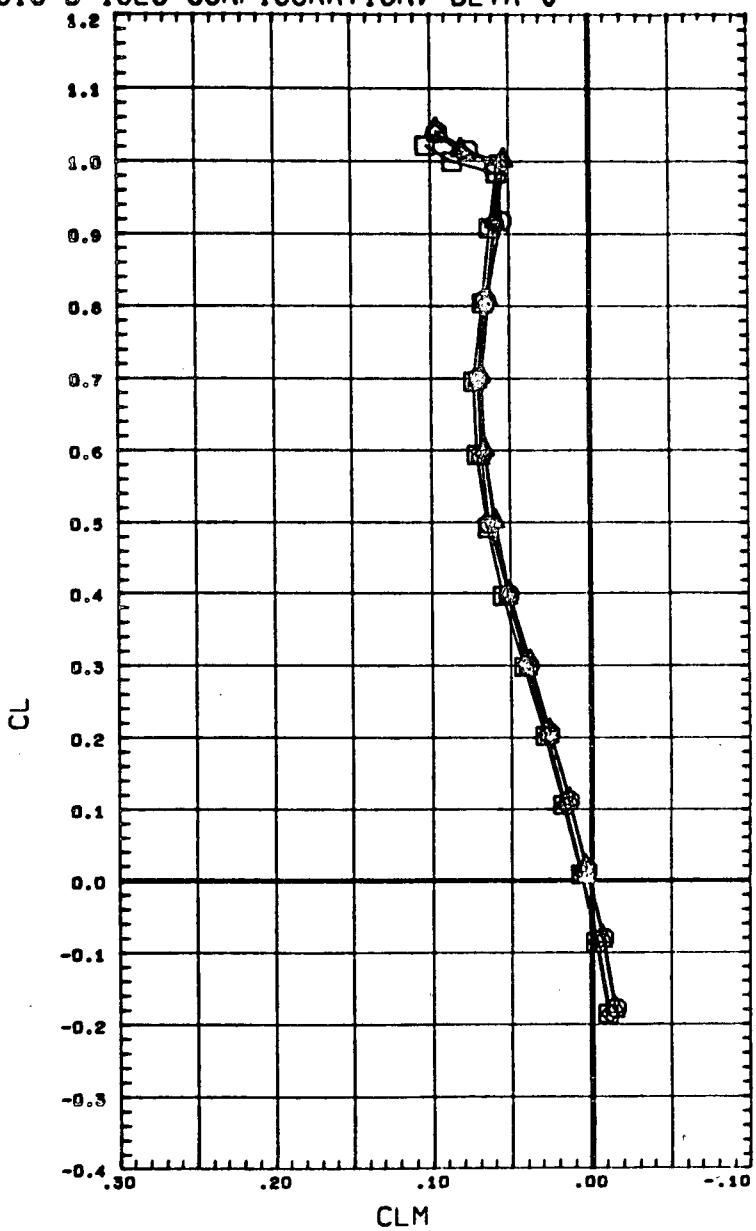
EFFECT OF SPLIT AILERON DEFLECTION- BASIC B-18E3 CONFIGURATION,  $\beta=0$



DATA SET SYMBOL CONFIGURATION DESCRIPTION

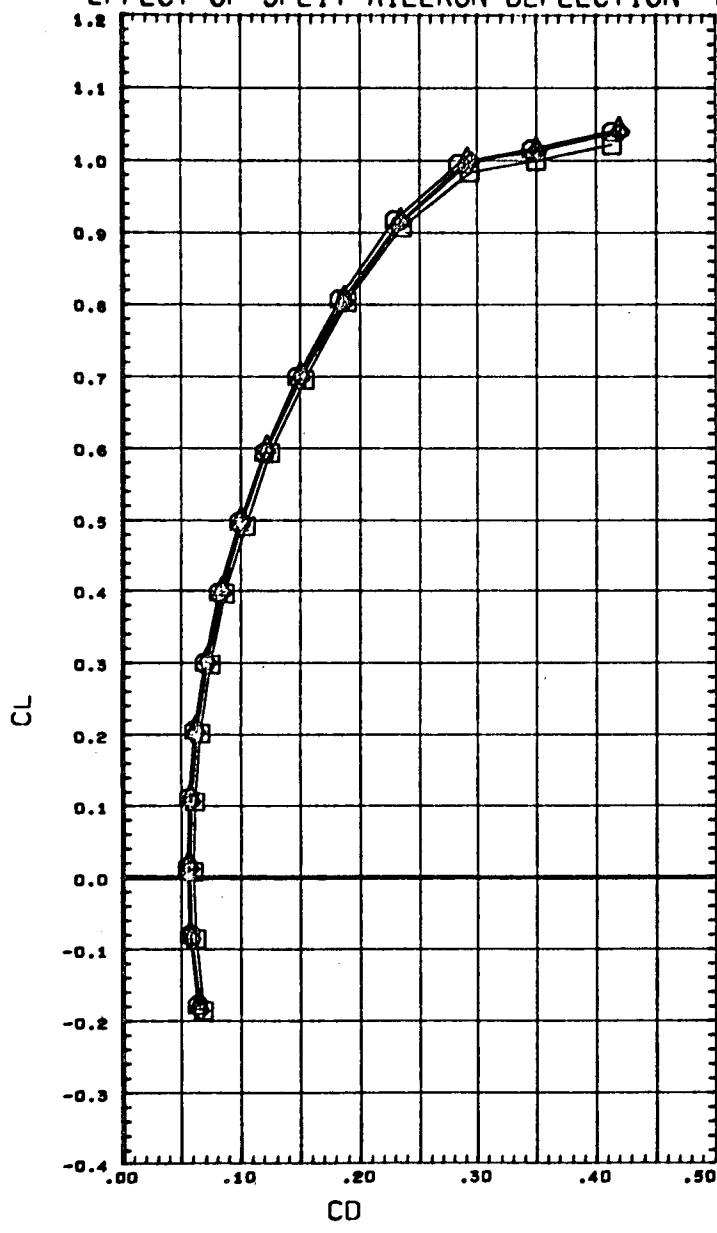
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 (AD6036) A GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)  
 (AD6034) D GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)  
 (AD6032) C GDLST 603-0 B30W23C10V14E38

MACH 0.201



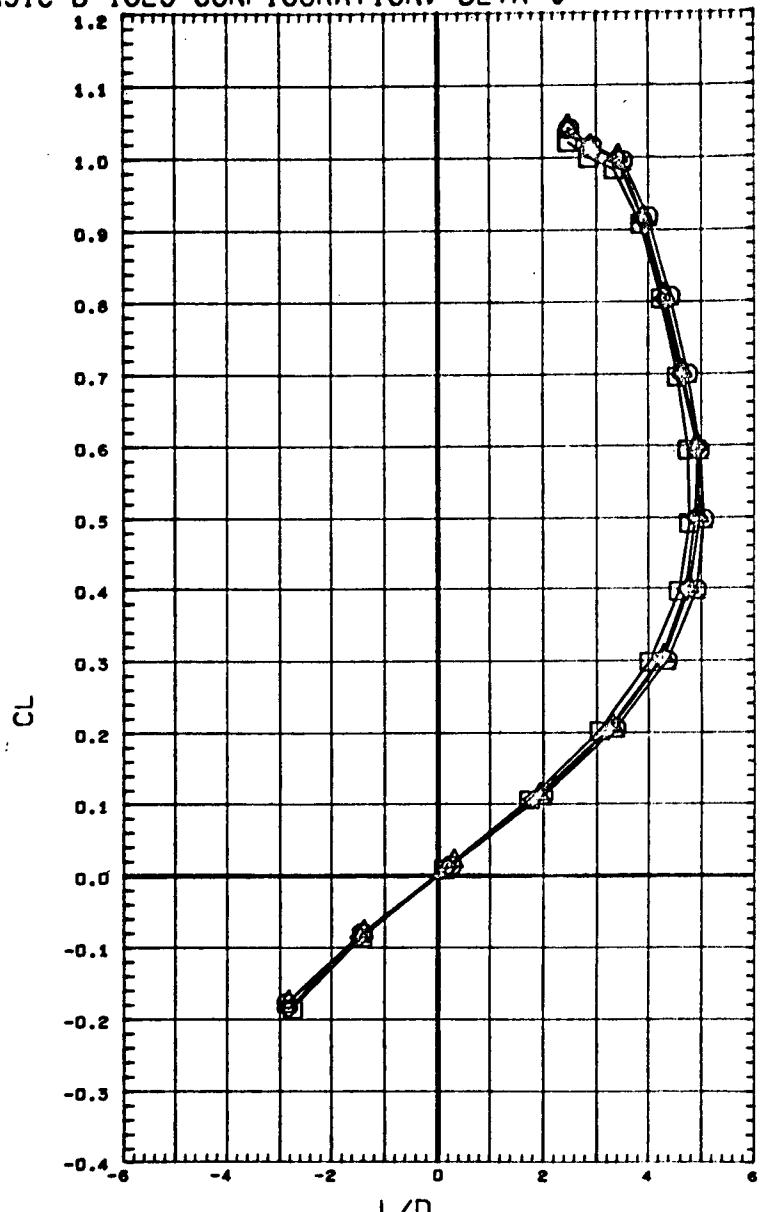
ELEVTR	CANARD	RUDDER	AILRON	REFERENCE	INFORMATION
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0.000	0.000	0.000	10.000	XMRP	29.0780 IN.
				YMRP	0.0000 IN.
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# EFFECT OF SPLIT AILERON DEFLECTION- BASIC B-18E3 CONFIGURATION, BETA=0



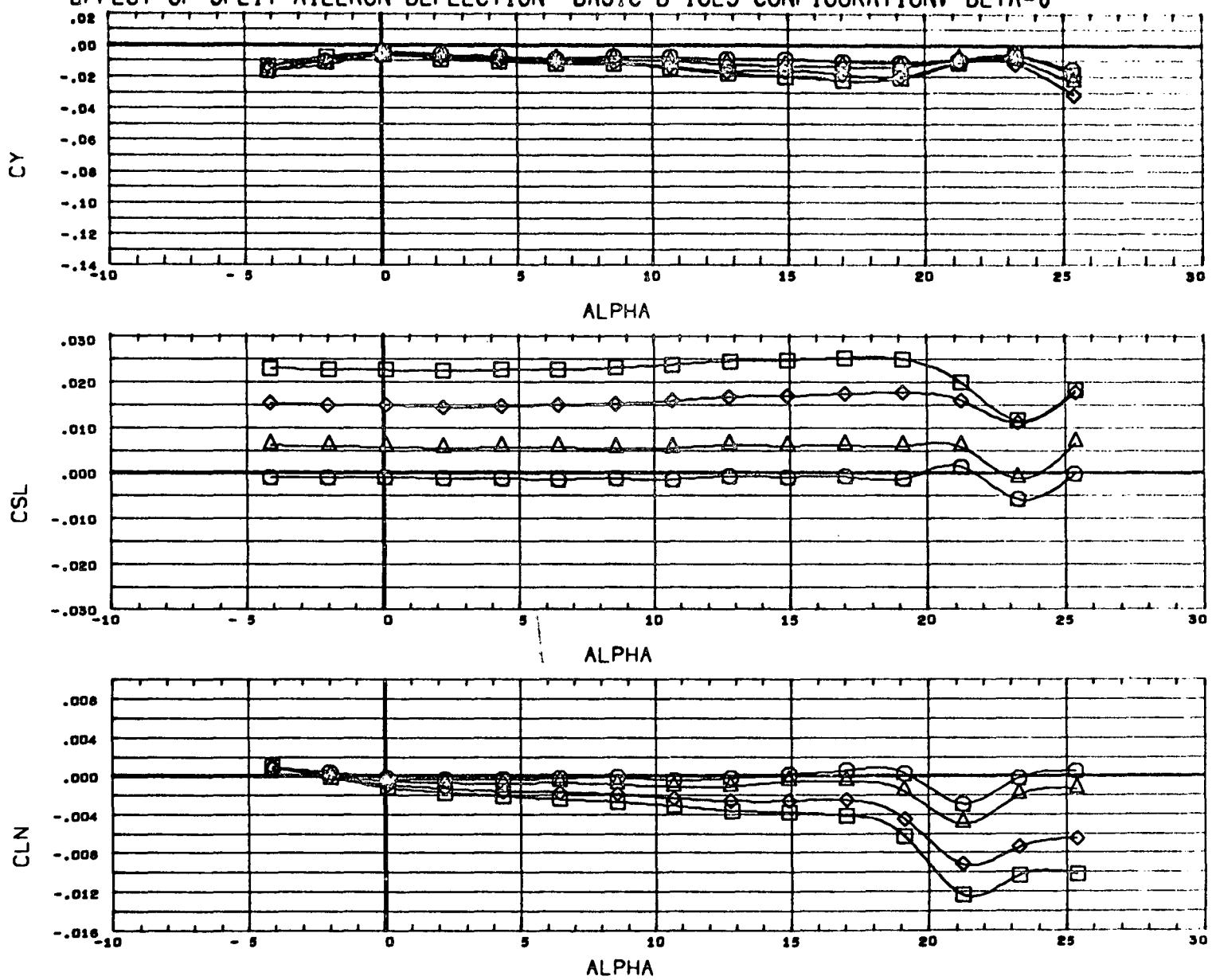
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6002) (O) GDLST 603-0 B30W23C10V14E38  
 (AD6036) (X) GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)  
 (AD6034) (D) GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)  
 (AD6032) (□) GDLST 603-0 B30W23C10V14E38

MACH 0.201



ELEVTR	CANARD	RUDDER	AILRON	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN.
0.000	0.000	0.000	3.330	LREF 16.1680 IN.
0.000	0.000	0.000	6.670	BREF 34.6320 IN.
0.000	0.000	0.000	10.000	XMRP 29.0780 IN.
				YMRP 0.0000 IN.
				ZMRP 0.0000 IN.
				SCALE 0.0200

EFFECT OF SPLIT AILERON DEFLECTION- BASIC B-18E3 CONFIGURATION,  $\beta=0$

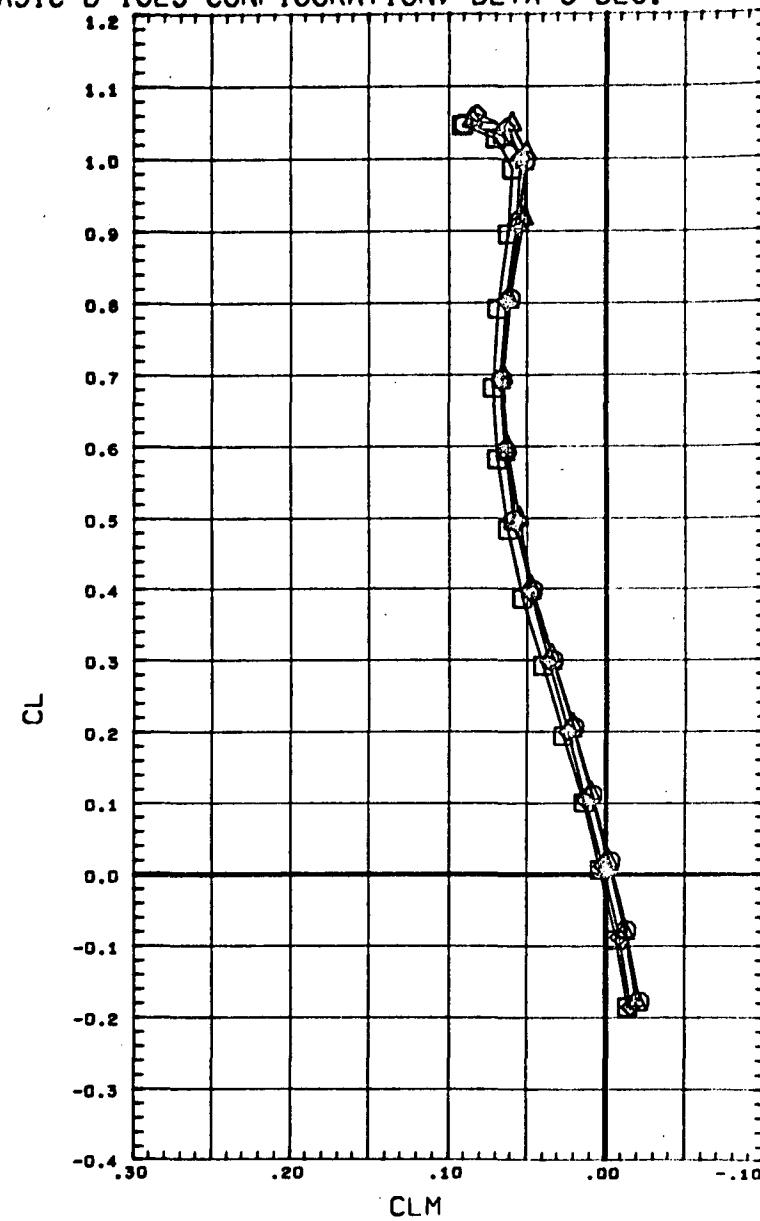
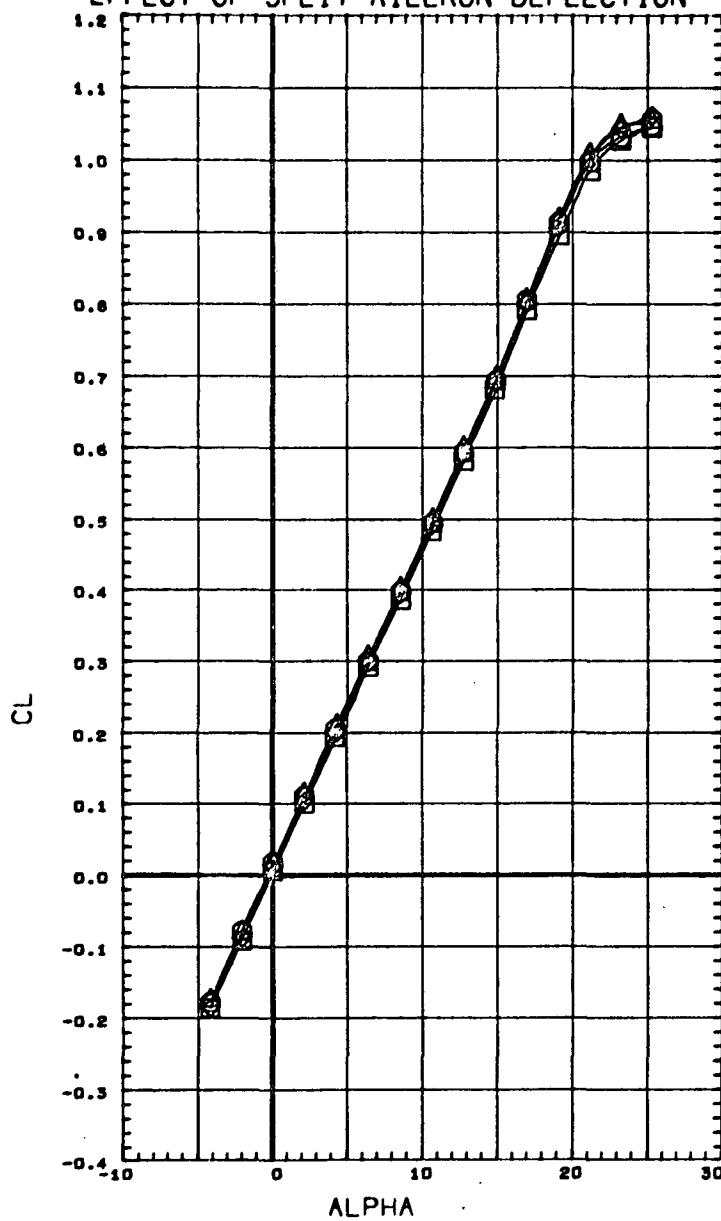


DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ELEVTR	CANARD	RUDDER	AILRON	REFERENCE	INFORMATION
(AD6002)	GDLST 603-0 B30W23C10V14E38	0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN
(AD6036)	GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)	0.000	0.000	0.000	3.330	LREF	16.1880 IN.
(AD6034)	GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)	0.000	0.000	0.000	6.670	BREF	34.6320 IN.
(AD6032)	GDLST 603-0 B30W23C10V14E38	0.000	0.000	0.000	10.000	XNRP	29.0780 IN.
						YMRP	0.0000 IN.
						ZHRP	0.0000 IN.
						SCALE	0.0200

MACH 0.201

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EFFECT OF SPLIT AILERON DEFLECTION- BASIC B-18E3 CONFIGURATION, BETA=5 DEG.

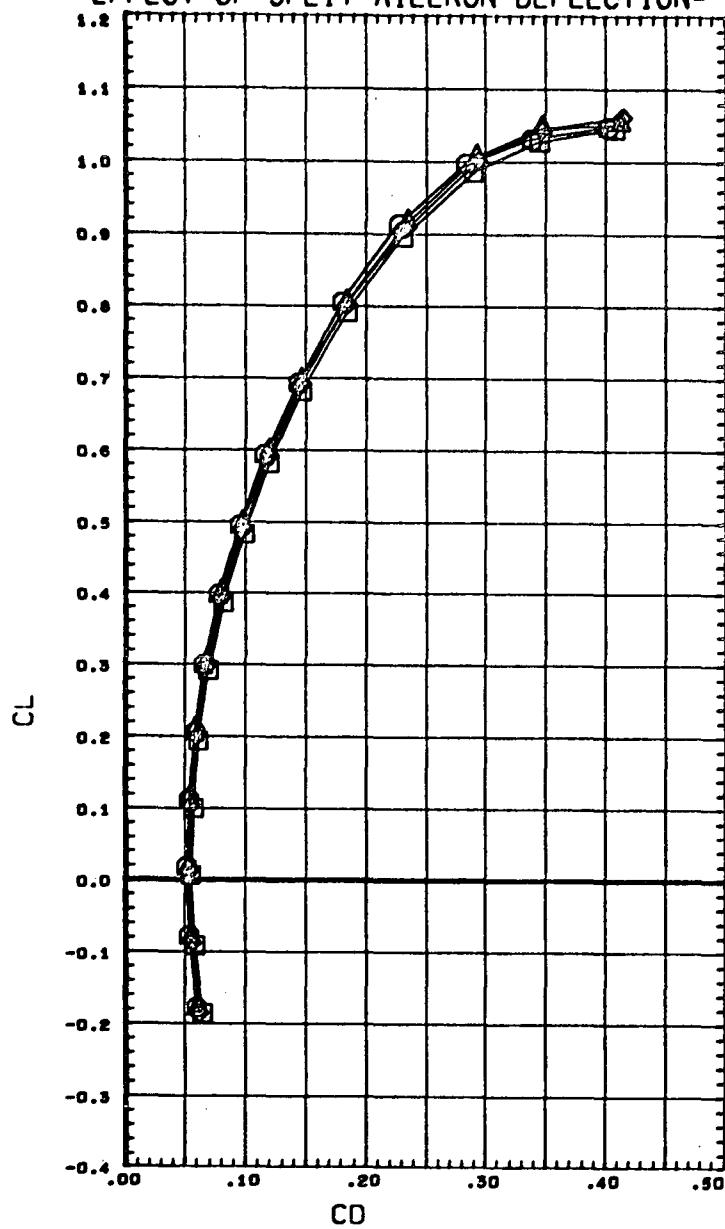


DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6003) GDLST 603-0 B30W23C10V14E38  
 (AD6037) GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)  
 (AD6035) GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)  
 (AD6033) GDLST 603-0 B30W23C10V14E38

MACH 0.201

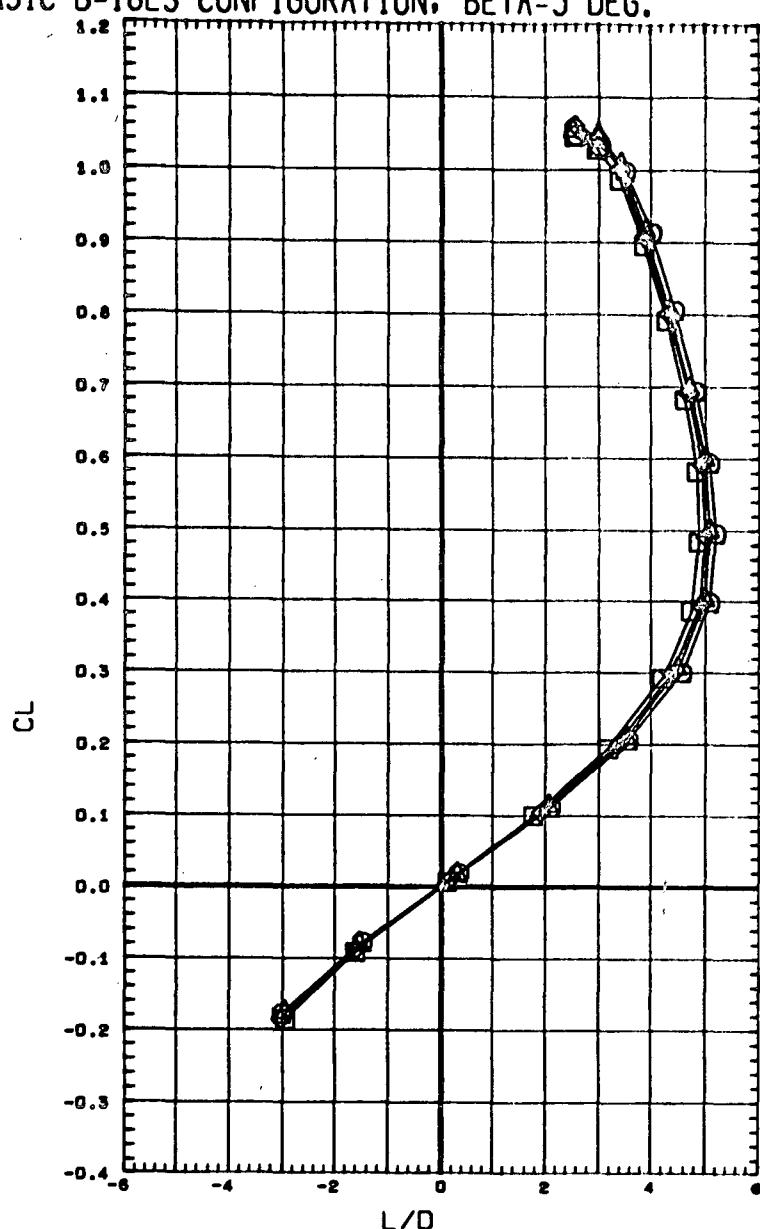
ELEVTR	CANARD	RUDER	AILRON	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN.
0.000	0.000	0.000	3.330	LREF 16.1880 IN.
0.000	0.000	0.000	6.670	BREF 34.6320 IN.
0.000	0.000	0.000	10.000	XMRP 29.0780 IN.
				YMRP 0.0000 IN.
				ZMRP 0.0000 IN.
				SCALE 0.0200

EFFECT OF SPLIT AILERON DEFLECTION- BASIC B-18E3 CONFIGURATION, BETA=5 DEG.



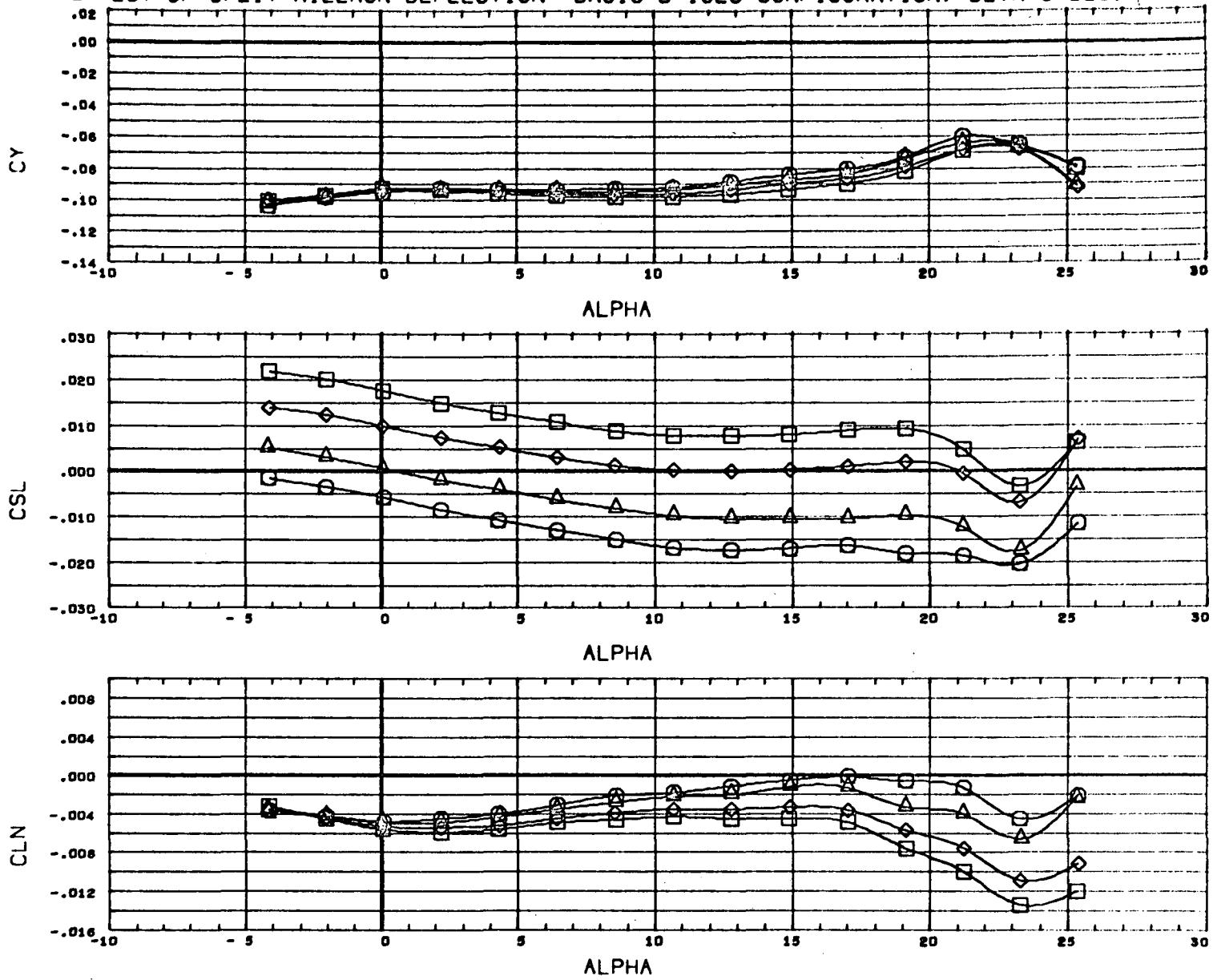
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6003) GDLST 603-0 B30W23C10V14E38  
 (AD6037) GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)  
 (AD6035) GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)  
 (AD6033) GDLST 603-0 B30W23C10V14E38

MACH 0.201



ELEVTR	CANARD	RUDDER	AILRDN	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN.
0.000	0.000	0.000	3.330	LREF	16.1880 IN.
0.000	0.000	0.000	6.670	BREF	34.6320 IN.
0.000	0.000	0.000	10.000	XNRP	29.0780 IN.
				YNRP	0.0000 IN.
				ZNRP	0.0000 IN.
				SCALE	0.0200

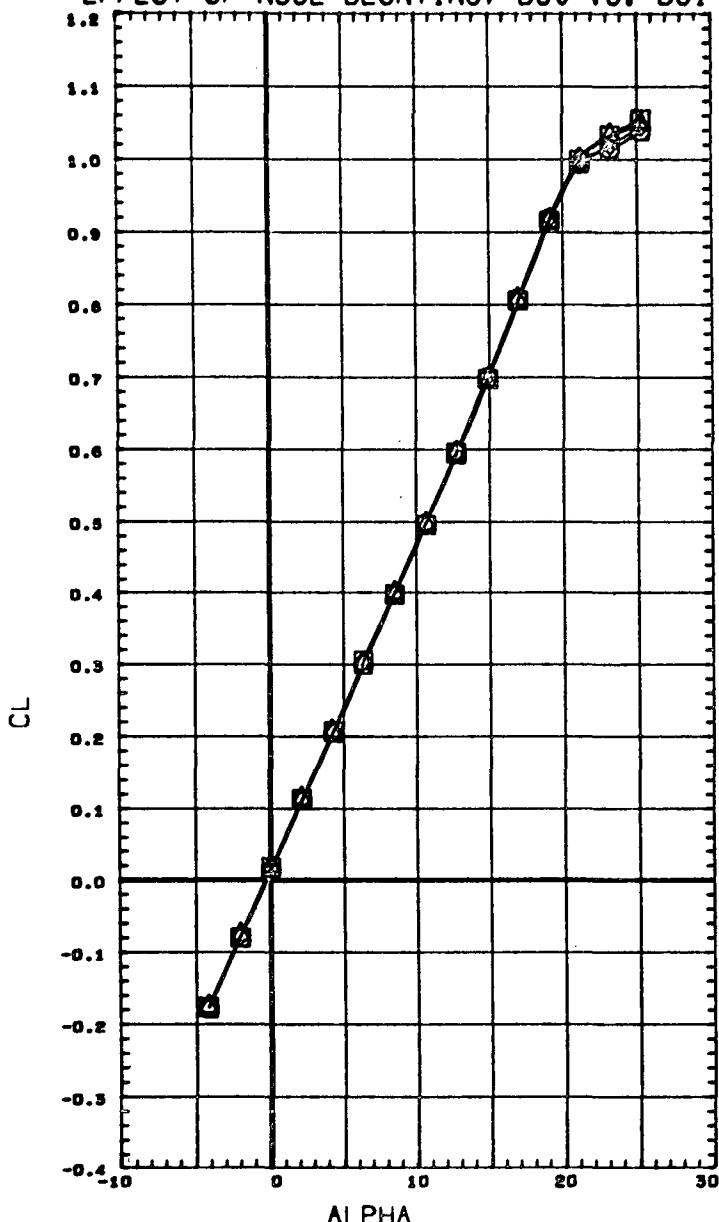
EFFECT OF SPLIT AILERON DEFLECTION- BASIC B-18E3 CONFIGURATION, BETA=5 DEG.



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ELEVTR	CANARD	RUDDER	AILRON	REFERENCE	INFORMATION
(AD603)	GDLST 603-0 B30W23C10V14E38	0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN
(AD6037)	GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)	0.000	0.000	0.000	3.330	LREF	16.1880 IN.
(AD6035)	GDLST 603-0 B30W23C10V14E38 (SPLIT ELEVONS)	0.000	0.000	0.000	6.670	BREF	34.6320 IN.
(AD6033)	GDLST 603-0 B30W23C10V14E38	0.000	0.000	0.000	10.000	XMRP	29.0780 IN.
						YMRP	0.0000 IN.
						ZMRP	0.0000 IN.
						SCALE	0.0200

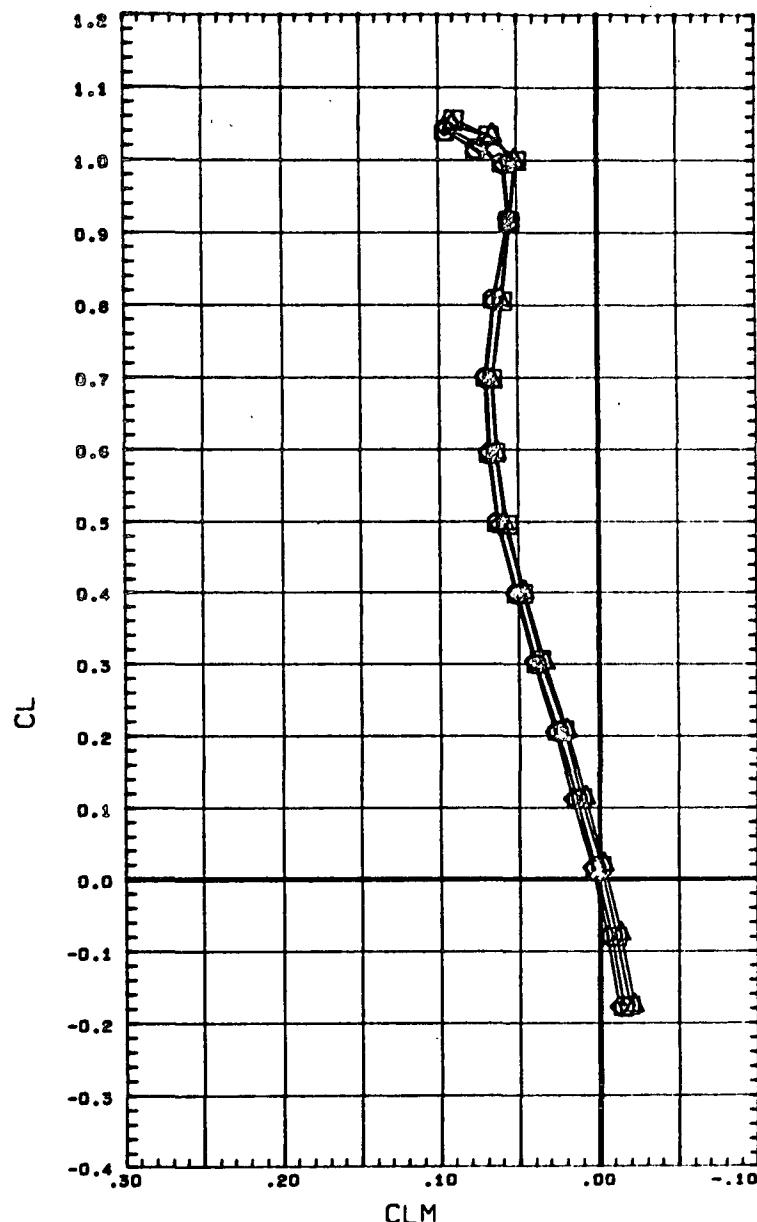
MACH 0.201

# EFFECT OF NOSE BLUNTING, B30 VS. B31



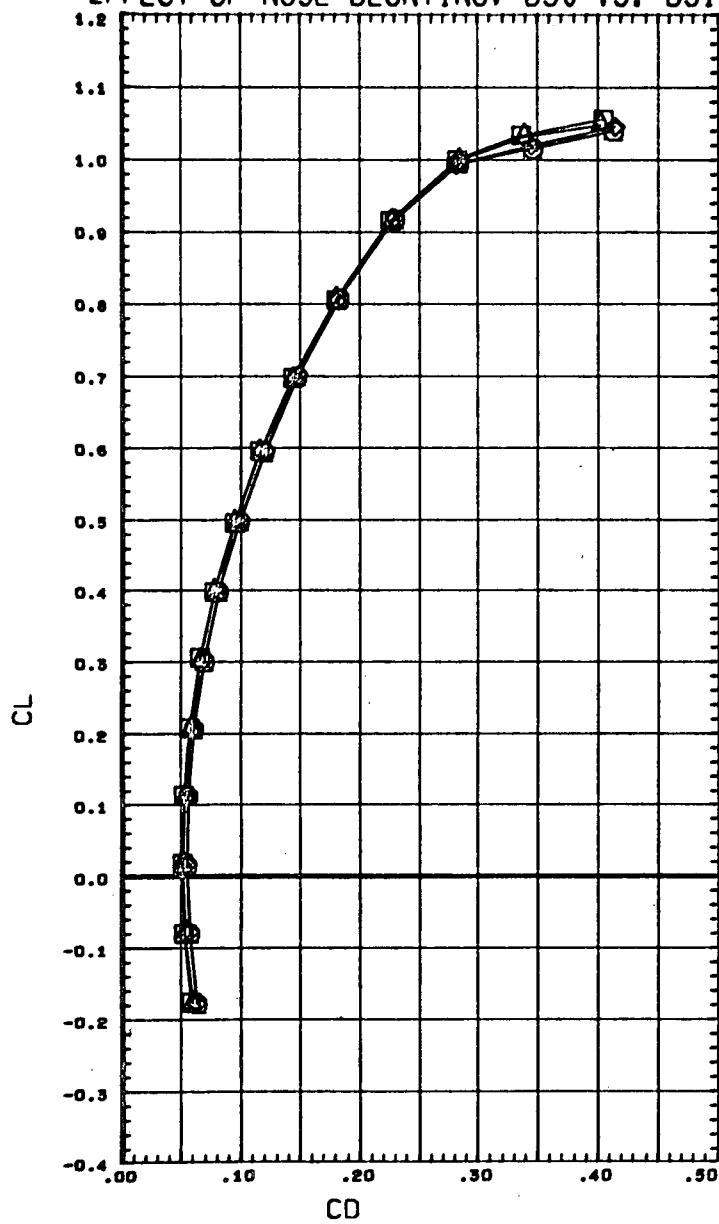
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6002) GDLST 603-0 B30W23C10V14E38  
 (AD6003) GDLST 603-0 B30W23C10V14E38  
 (AD6015) GDLST 603-0 B31W23C10V14E38  
 (AD6016) GDLST 603-0 B31W23C10V14E38

MACH 0.201



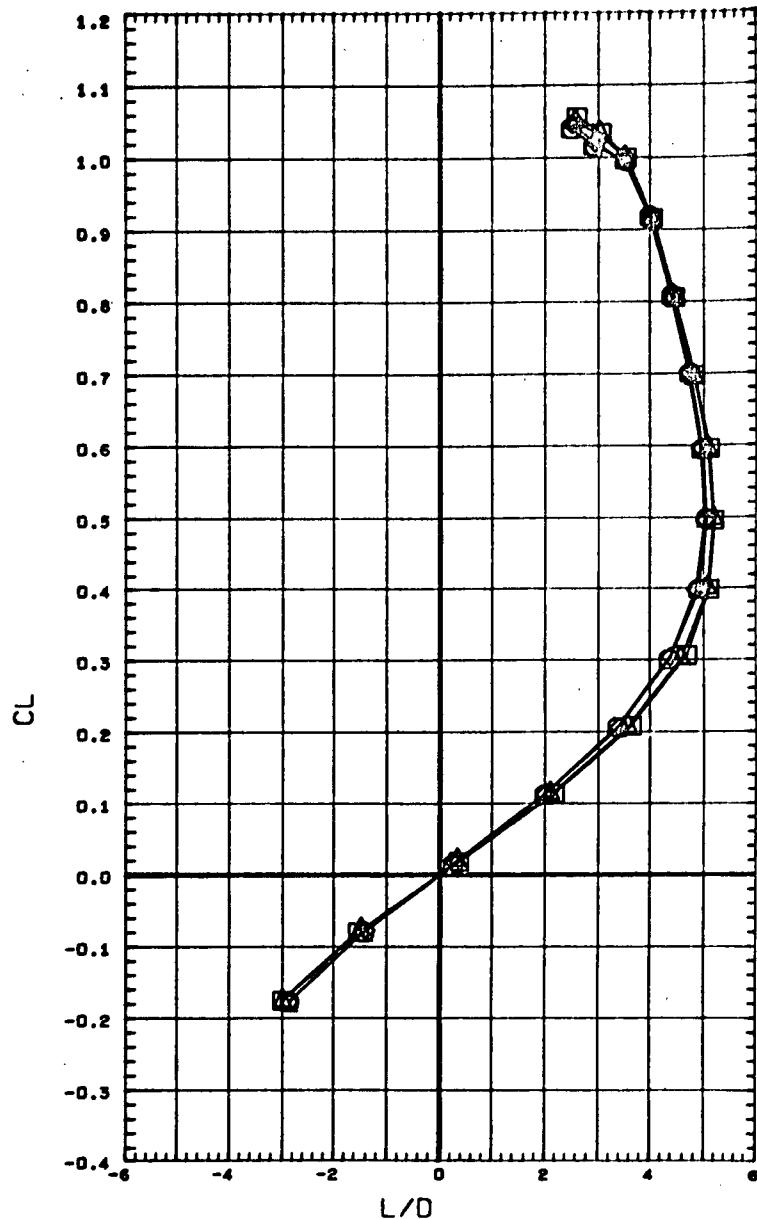
BETA ELEVTR CANARD AILRDN REFERENCE INFORMATION  
 0.000 0.000 0.000 0.000 SREF 492.4804 SQ. IN.  
 5.000 0.000 0.000 0.000 LREF 16.1880 IN.  
 0.000 0.000 0.000 0.000 BREF 34.6320 IN.  
 5.000 0.000 0.000 0.000 XMRP 29.0780 IN.  
 YMRP 0.0000 IN.  
 ZMRP 0.0000 IN.  
 SCALE 0.0200

## EFFECT OF NOSE BLUNTING, B30 VS. B31



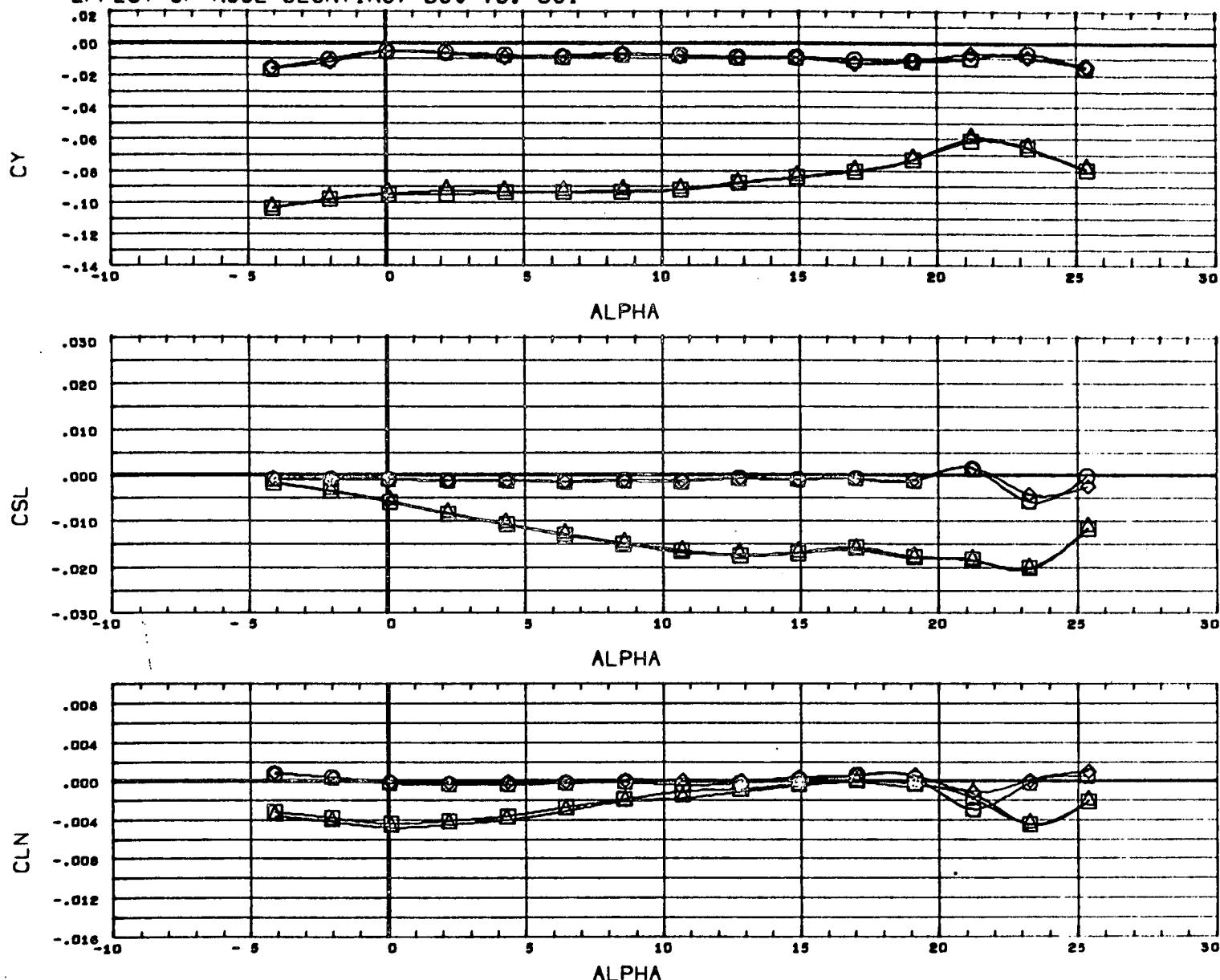
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6002) GDLST 603-D B30W23C10V14E38  
 (AD6003) GDLST 603-D B30W23C10V14E38  
 (AD6015) GDLST 603-D B31W23C10V14E38  
 (AD6016) GDLST 603-D B31W23C10V14E38

MACH 0.201



BETA	ELEVTR	CANARD	AILRON	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN.
5.000	0.000	0.000	0.000	LREF 16.1680 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
5.000	0.000	0.000	0.000	XMRP 29.0780 IN.
				YMRP 0.0000 IN.
				ZMRP 0.0000 IN.
				SCALE 0.0200

## EFFECT OF NOSE BLUNTING, B30 VS. B31

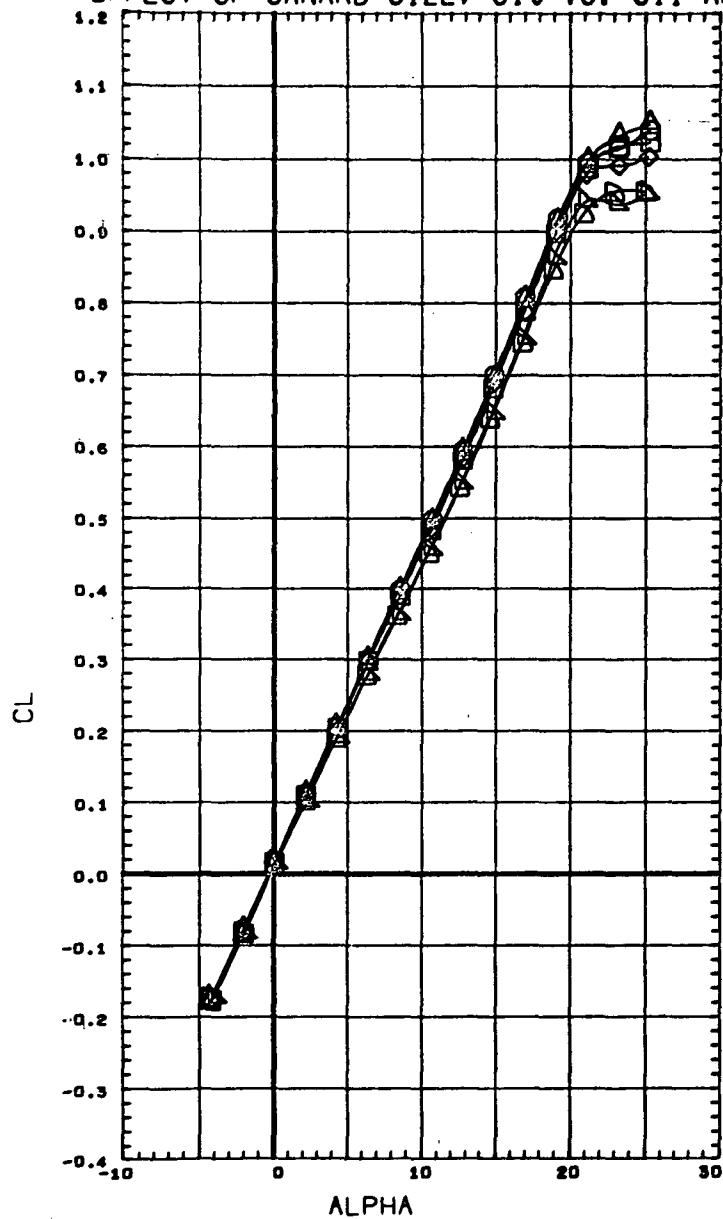


DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6002) GDLST 603-0 B30W23C10V14E38  
 (AD6003) GDLST 603-0 B30W23C10V14E38  
 (AD6015) GDLST 603-0 B31W23C10V14E38  
 (AD6016) GDLST 603-0 B31W23C10V14E38

BETA	ELEVTR	RUDDER	AILRON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN
5.000	0.000	0.000	0.000	LREF	16.1880 IN.
0.000	0.000	0.000	0.000	BREF	34.6320 IN.
5.000	0.000	0.000	0.000	XMRP	29.0780 IN.
				YMRP	0.0000 IN.
				ZMRP	0.0000 IN.
				SCALE	0.0200

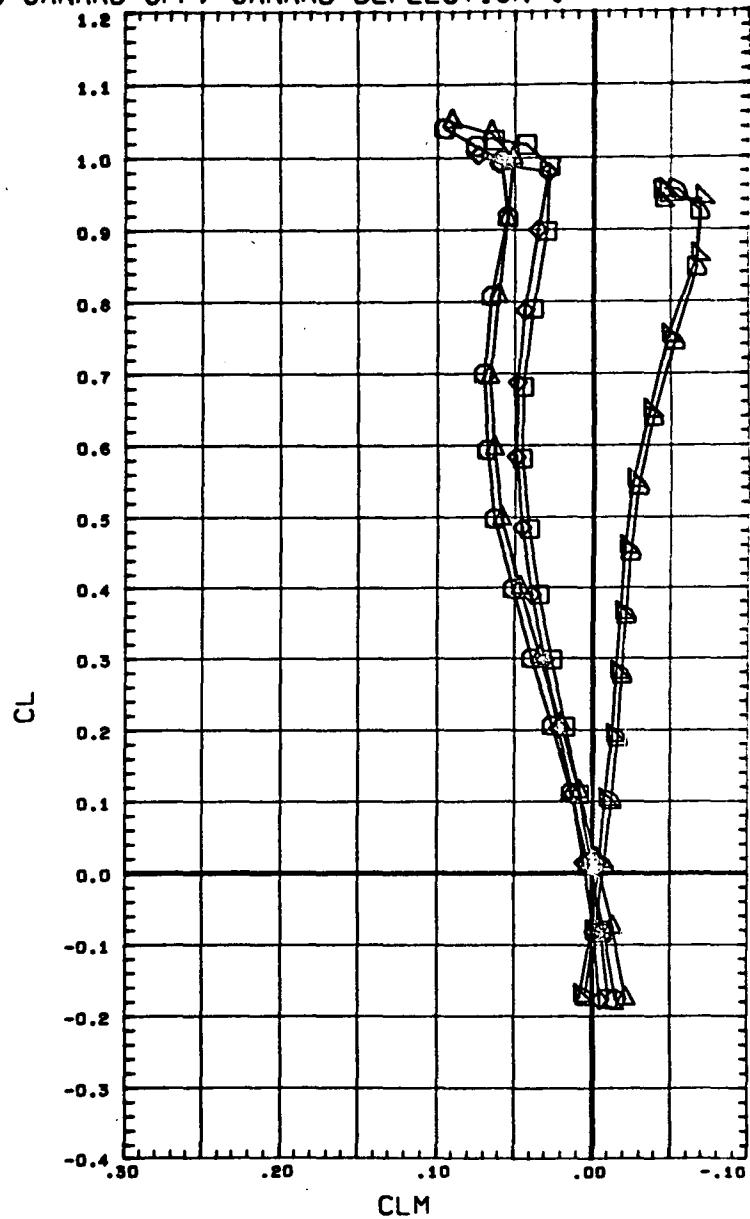
MACH 0.201

## EFFECT OF CANARD SIZE, C10 VS. C11 AND CANARD OFF, CANARD DEFLECTION=0



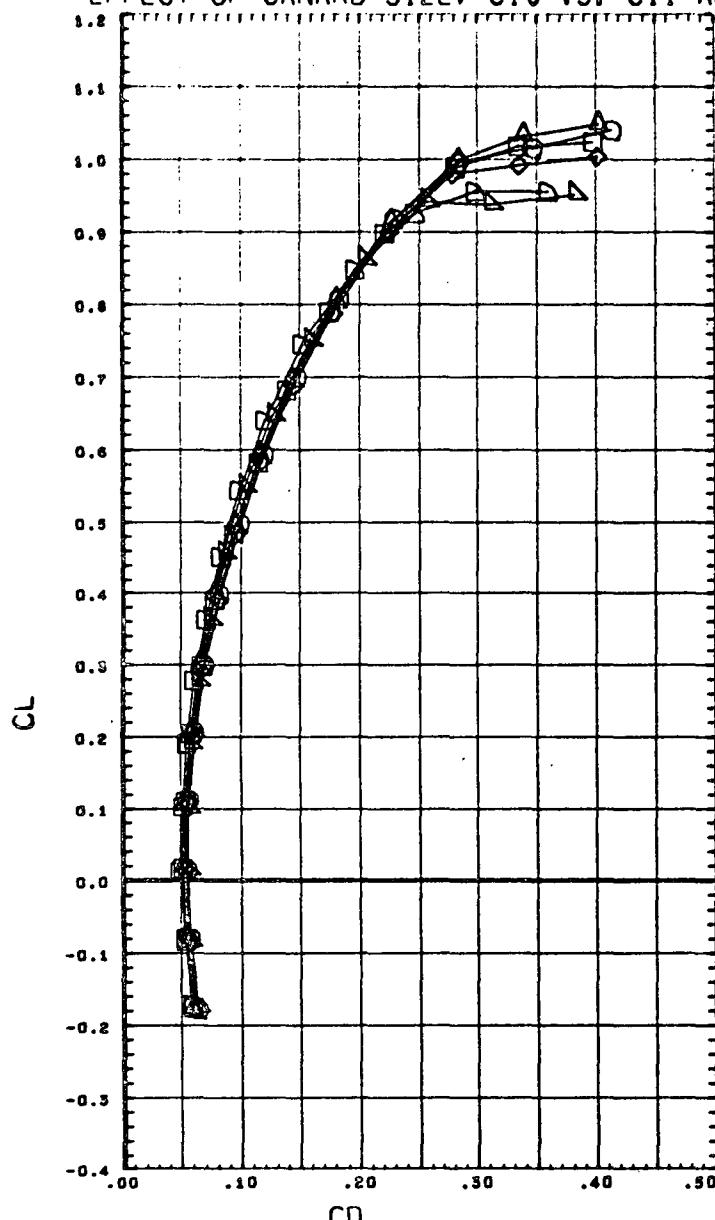
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6002) GDLST 603-0 B30W23C10V14E38  
 (AD6003) GDLST 603-0 B30W23C10V14E38  
 (AD6046) GDLST 603-0 B30W23C11V14E38  
 (AD6047) GDLST 603-0 B30W23C11V14E38  
 (AD6044) GDLST 603-0 B30W23V14E38  
 (AD6045) GDLST 603-0 B30W23V14E38

MACH 0.201



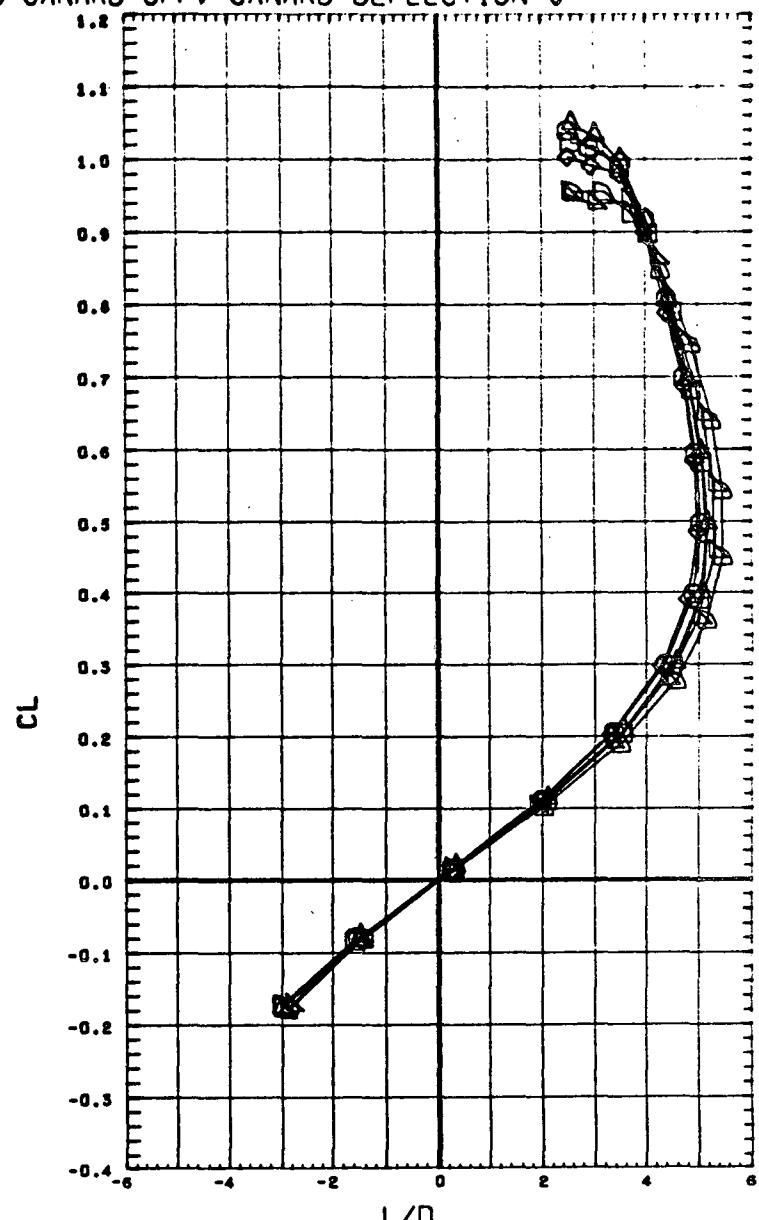
BETA	ELEVTR	RUDDER	AIRLON	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN
5.000	0.000	0.000	0.000	LREF 16.1880 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
5.000	0.000	0.000	0.000	XMRP 29.0780 IN.
0.000	0.000	0.000	0.000	YMRP 0.0000 IN.
5.000	0.000	0.000	0.000	ZMRP 0.0000 IN.
				SCALE 0.0200

# EFFECT OF CANARD SIZE, C10 VS. C11 AND CANARD OFF, CANARD DEFLECTION=0



DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AD6052)	GCLST 603-0 B30W23C10V14E38
(AD6003)	GCLST 603-0 B30W23C10V14E38
(AD6046)	GCLST 603-0 B30W23C11V14E38
(AD6047)	GCLST 603-0 B30W23C11V14E38
(AD6044)	GCLST 603-0 B30W23V14E38
(AD6045)	GCLST 603-0 B30W23V14E38

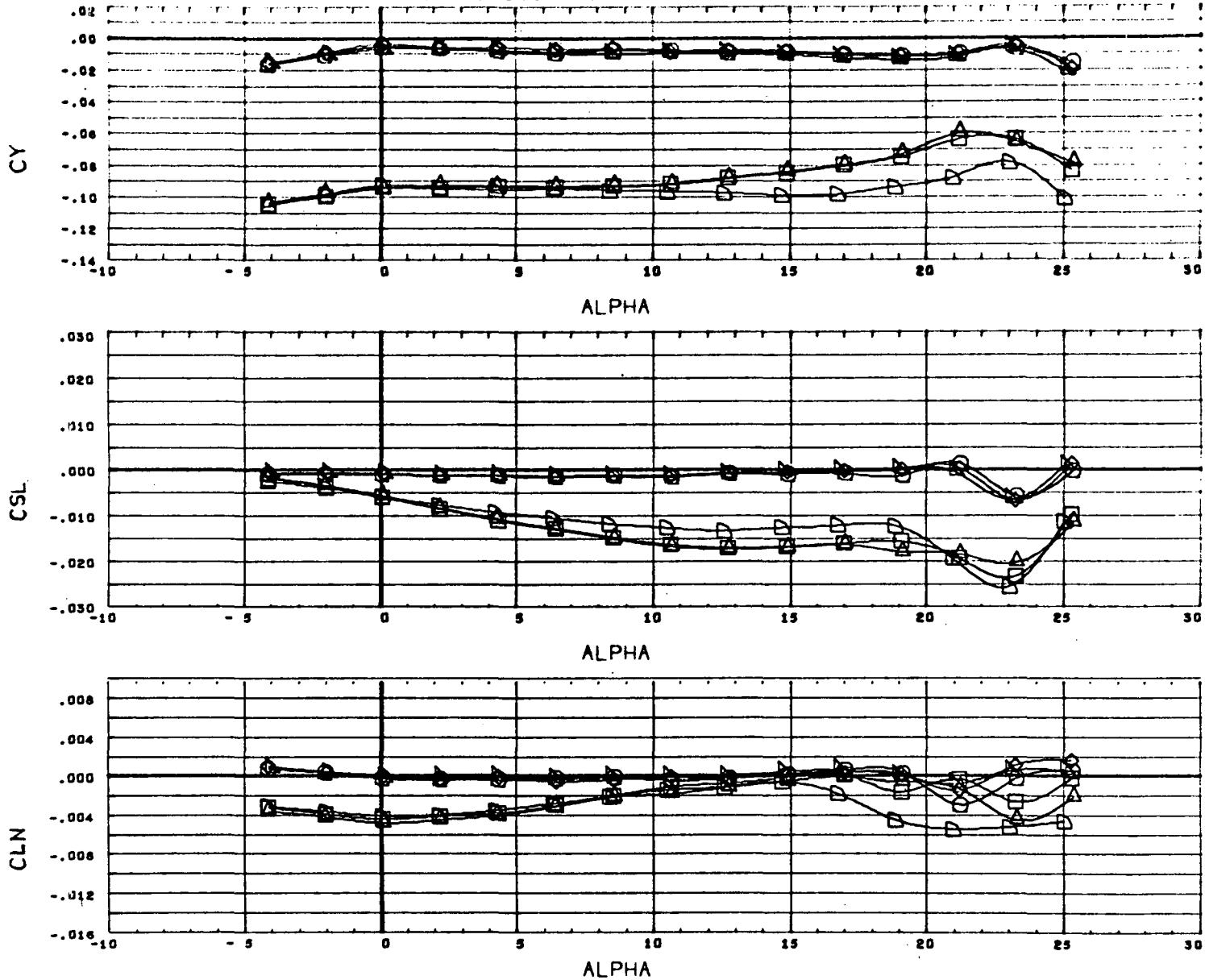
MACH .20



BETA	ELEVTR	RUDDER	AILRON	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4894 SQ. IN.
5.000	0.000	0.000	0.000	LREF 16.1880 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
5.000	0.000	0.000	0.000	XHRP 29.0780 IN.
0.000	0.000	0.000	0.000	YHRF 0.0000 IN.
5.000	0.000	0.000	0.000	ZHRP 0.0000 IN.
				SCALE 0.0200

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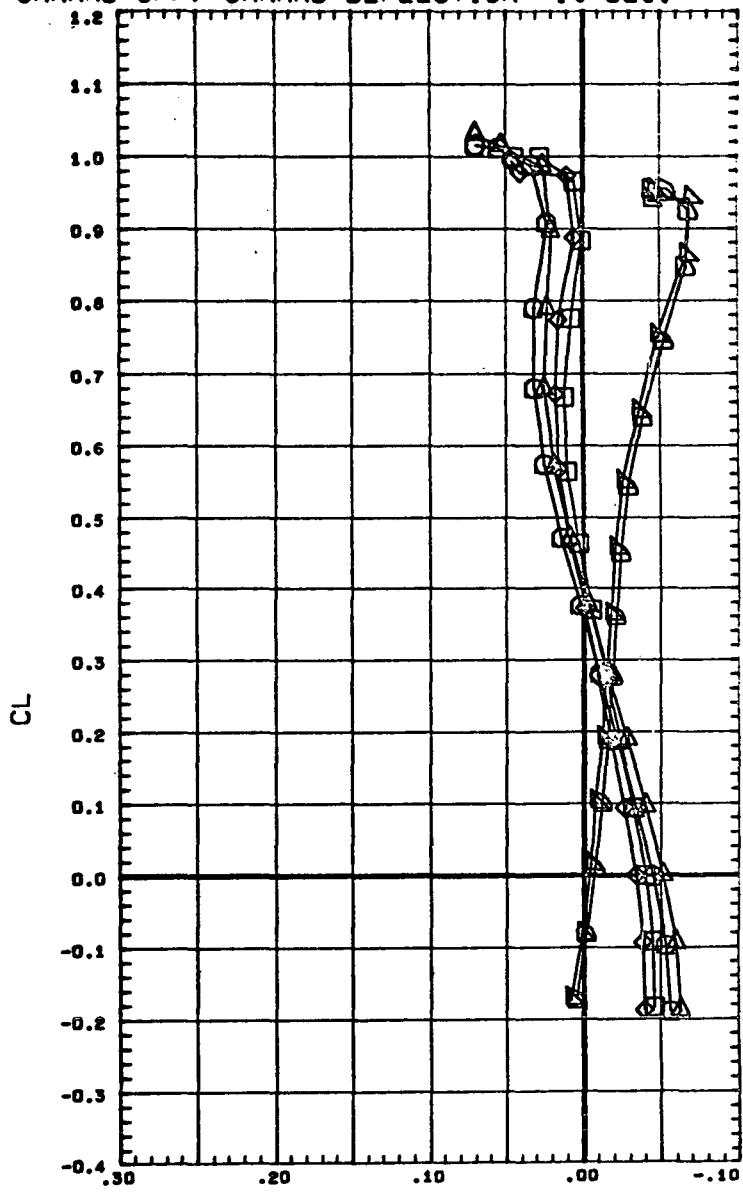
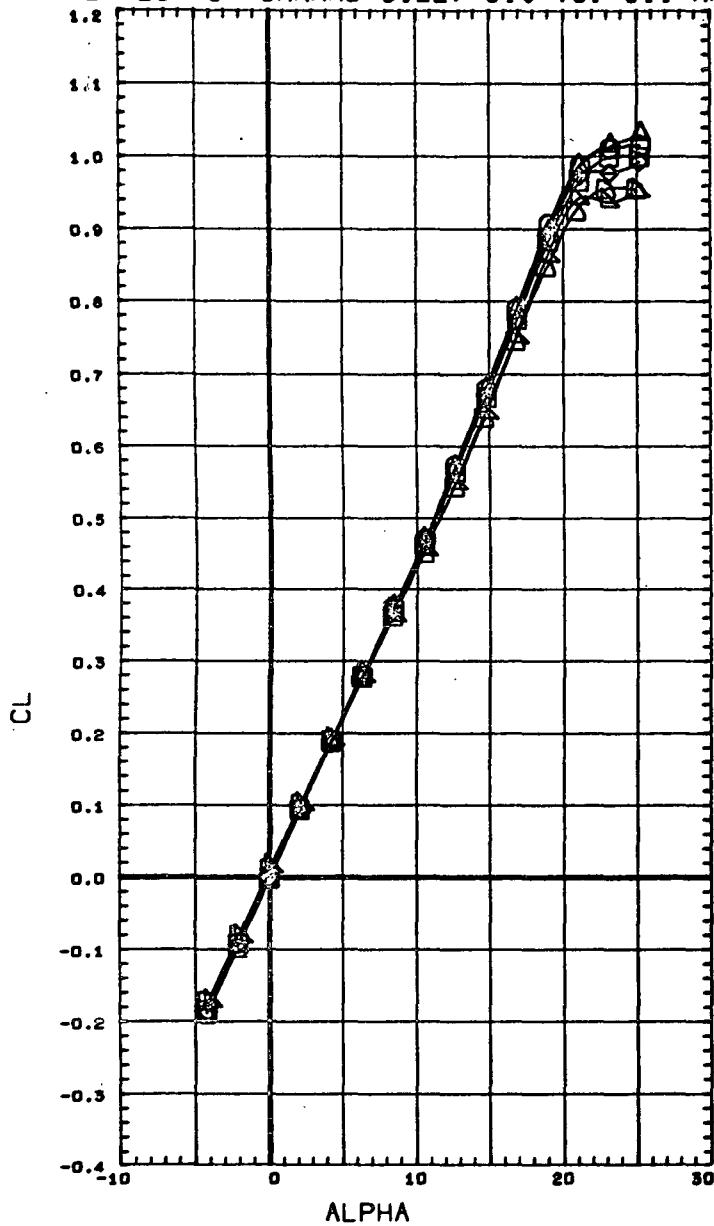
EFFECT OF CANARD SIZE, C10 VS. C11 AND CANARD OFF. CANARD DEFLECTION=0



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AC6002) GCLST 603-0 B30W23C10V14E38  
 (AC6003) GCLST 603-0 B30W23C10V14E38  
 (AC6046) GCLST 603-0 B30W23C11V14E38  
 (AC6047) GCLST 603-0 B30W23C11V14E38  
 (AC6044) GCLST 603-0 B30W23V14E38  
 (AC6045) GCLST 603-0 B30W23V14E38

BETA	ELEVTR	RUDER	AIRLON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4854
5.000	0.000	0.000	0.000	LREF	16.1880
0.000	0.000	0.000	0.000	BREF	34.6320
5.000	0.000	0.000	0.000	XMRP	29.0760
0.000	0.000	0.000	0.000	YMRP	0.0000
5.000	0.000	0.000	0.000	ZMRP	0.0000
				SCALE	0.0200

EFFECT OF CANARD SIZE, C10 VS. C11 AND CANARD OFF, CANARD DEFLECTION=-10 DEG.

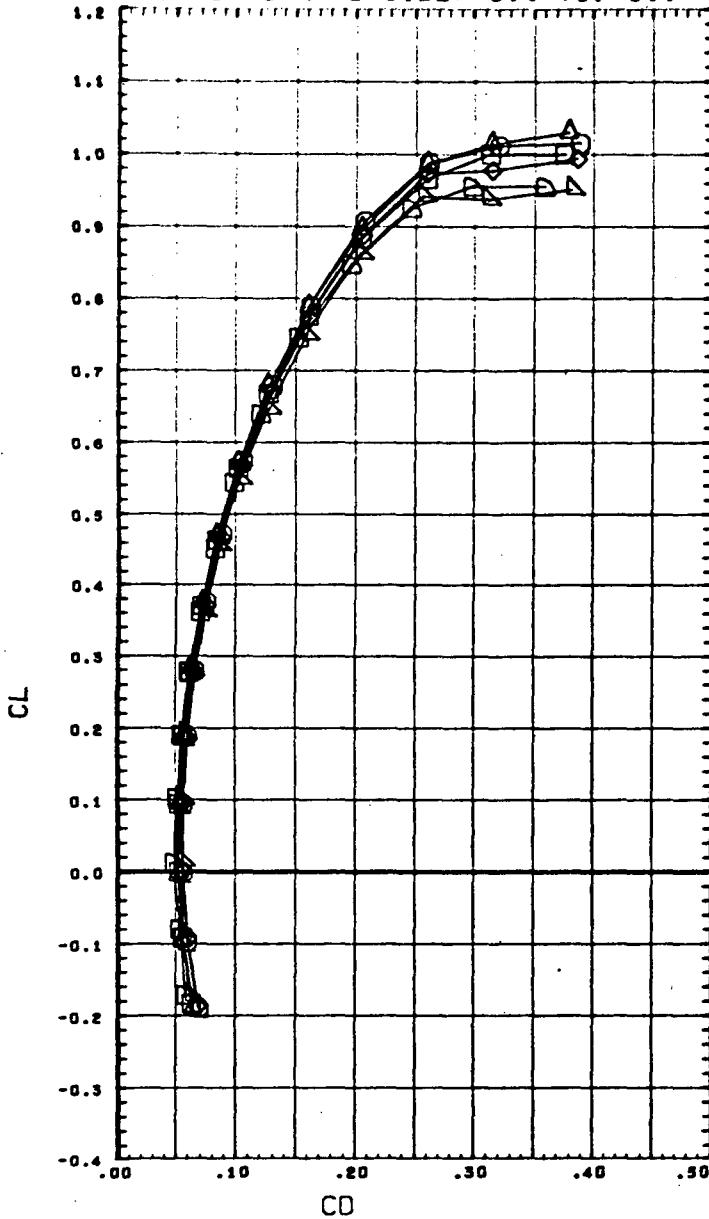


DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6009) DGD09 GDLST 603-0 B30W23C10V14E38  
 (AD6010) DGD010 GDLST 603-0 B30W23C10V14E38  
 (AD6050) DGD050 GDLST 603-0 B30W23C11V14E38  
 (AD6051) DGD051 GDLST 603-0 B30W23C11V14E38  
 (AD6244) DGD244 GDLST 603-0 B30W23V14E38  
 (AD6045) DGD045 GDLST 603-0 B30W23V14E38

MACH 0.201

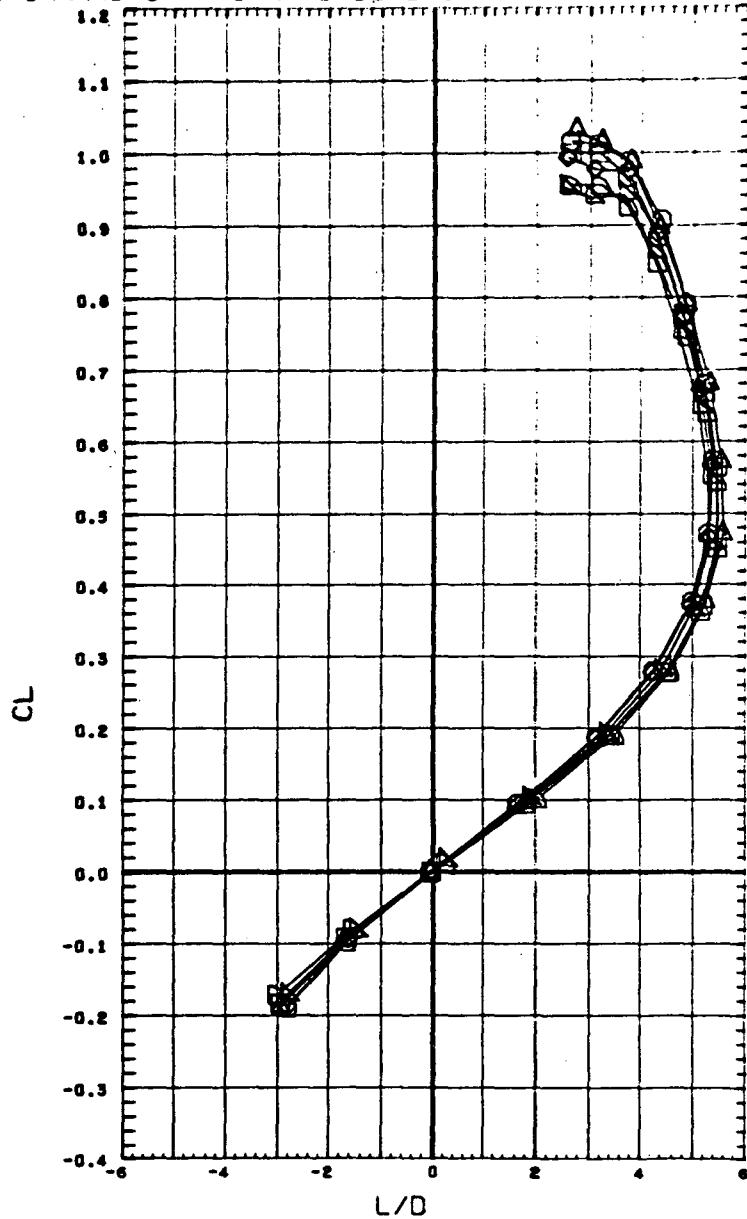
BETA	ELEVTR	RUDDER	AILRON	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN
5.000	0.000	0.000	0.000	LREF 16.1980 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
5.000	0.000	0.000	0.000	XHRP 29.0780 IN.
0.000	0.000	0.000	0.000	YHRP 0.0000 IN.
5.000	0.000	0.000	0.000	ZHRP 0.0000 IN.
				SCALE 0.0200

EFFECT OF CANARD SIZE, C10 VS. C11 AND CANARD OFF. CANARD DEFLECTION=-10 DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AC6009) GCLST 603-0 830W23C10V14E38  
 (AC6010) GCLST 603-0 830W23C10V14E38  
 (AC6050) GCLST 603-0 830W23C11V14E38  
 (AD6051) GCLST 603-0 830W23C11V14E38  
 (AC6044) GCLST 603-0 830W23V14E38  
 (AC6045) GCLST 603-0 830W23V14E38

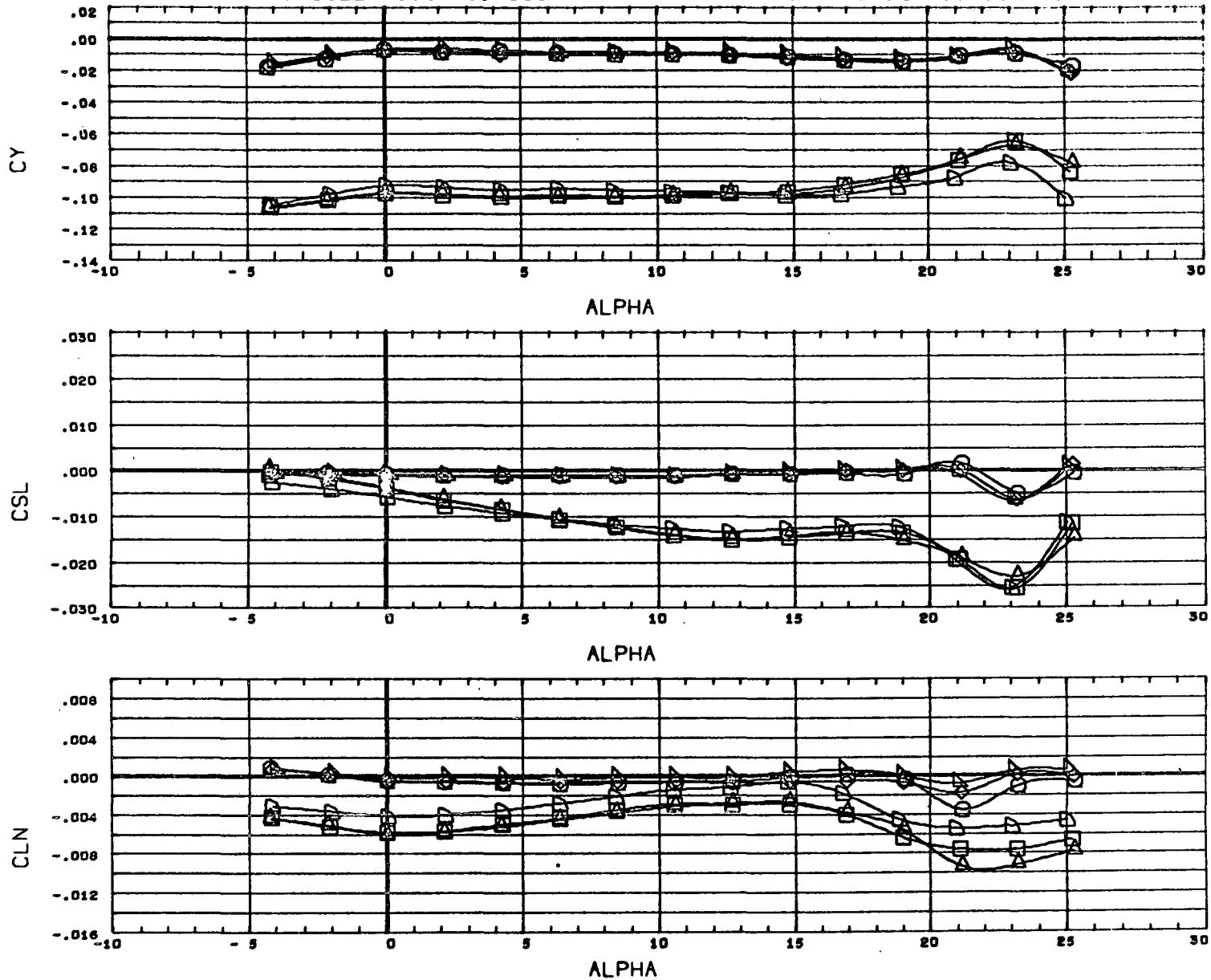
MACH .20



BETA	ELEVTR	RUDDER	AILRON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804
5.000	0.000	0.000	0.000	LREF	16.1889
0.000	0.000	0.000	0.000	BREF	34.6320
5.000	0.000	0.000	0.000	XHRP	29.0780
0.000	0.000	0.000	0.000	YHRP	0.0000
5.000	0.000	0.000	0.000	ZHRP	0.0000
				SCALE	0.0200

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EFFECT OF CANARD SIZE, C10 VS. C11 AND CANARD OFF, CANARD DEFLECTION=-10 DEG.

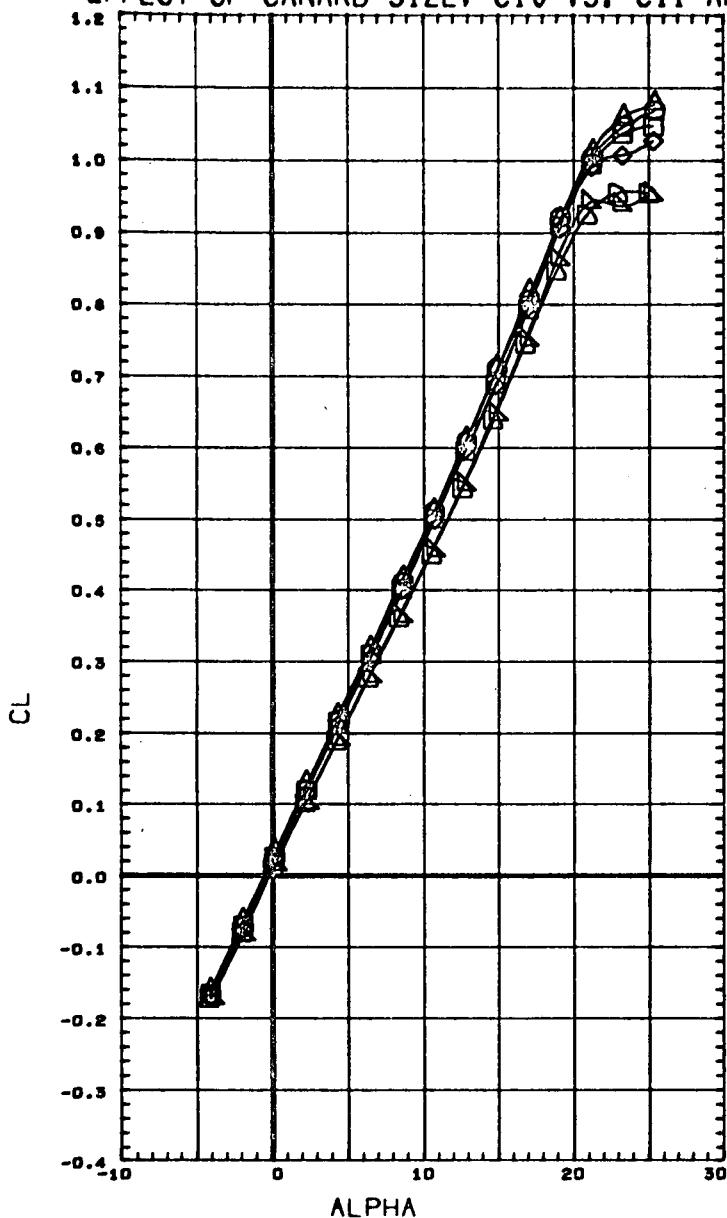


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AD6009)	GDLST 603-0 B30W23C10V14E38
(AD6010)	GDLST 603-0 B30W23C10V14E38
(AD6050)	GDLST 603-0 B30W23C11V14E38
(AD6051)	GDLST 603-0 B30W23C11V14E38
(AD6044)	GDLST 603-0 B30W23V14E38
(AD6045)	GDLST 603-0 B30W23V14E38

MACH 0.201

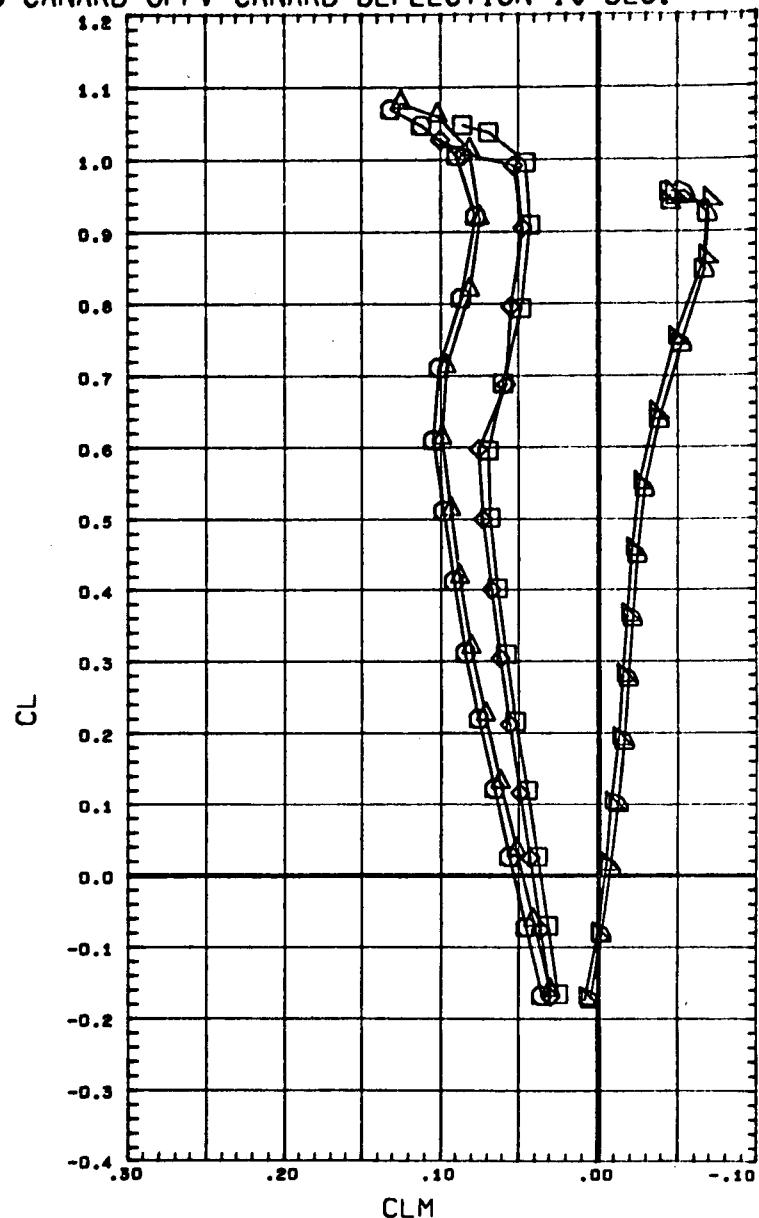
BETA	ELEVTR	RUDDER	AILRON	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN
5.000	0.000	0.000	0.000	LREF 16.1680 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
5.000	0.000	0.000	0.000	XMRP 29.0780 IN.
0.000	0.000	0.000	0.000	YMRP 0.0000 IN.
5.000	0.000	0.000	0.000	ZMRP 0.0000 IN.
				SCALE 0.0200

EFFECT OF CANARD SIZE, C10 VS. C11 AND CANARD OFF. CANARD DEFLECTION=10 DEG.



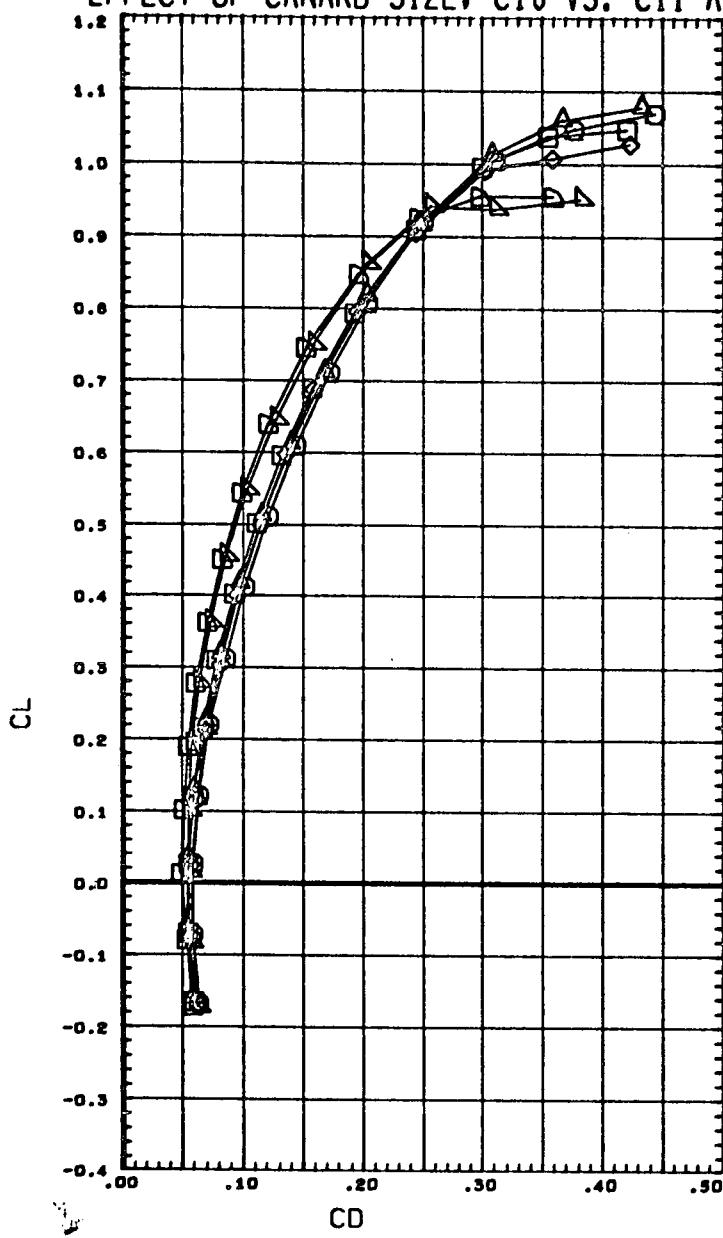
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AD6007)	GDLST 603-0 B30W23C10V14E38
(AD6008)	GDLST 603-0 B30W23C10V14E38
(AD6048)	GDLST 603-0 B30W23C11V14E38
(AD6049)	GDLST 603-0 B30W23C11V14E38
(AD6044)	GDLST 603-0 B30W23V14E38
(AD6045)	GDLST 603-0 B30W23V14E38

MACH 0.201



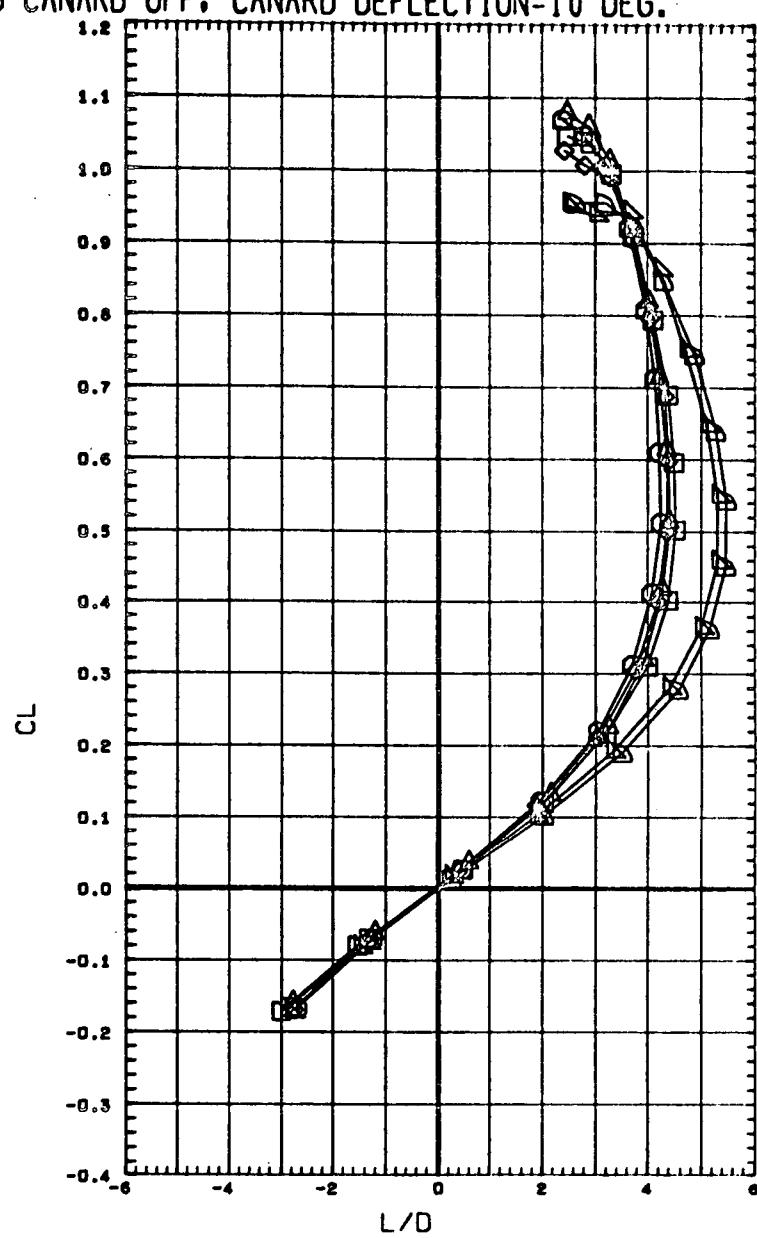
BETA	ELEVTR	RUDDER	AIRLON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN.
5.000	0.000	0.000	0.000	LREF	16.1880 IN.
0.000	0.000	0.000	0.000	BREF	34.6320 IN.
5.000	0.000	0.000	0.000	XMRP	29.0780 IN.
0.000	0.000	0.000	0.000	YMRP	0.0000 IN.
5.000	0.000	0.000	0.000	ZMRP	0.0000 IN.
				SCALE	0.0200

EFFECT OF CANARD SIZE, C10 VS. C11 AND CANARD OFF, CANARD DEFLECTION=10 DEG.



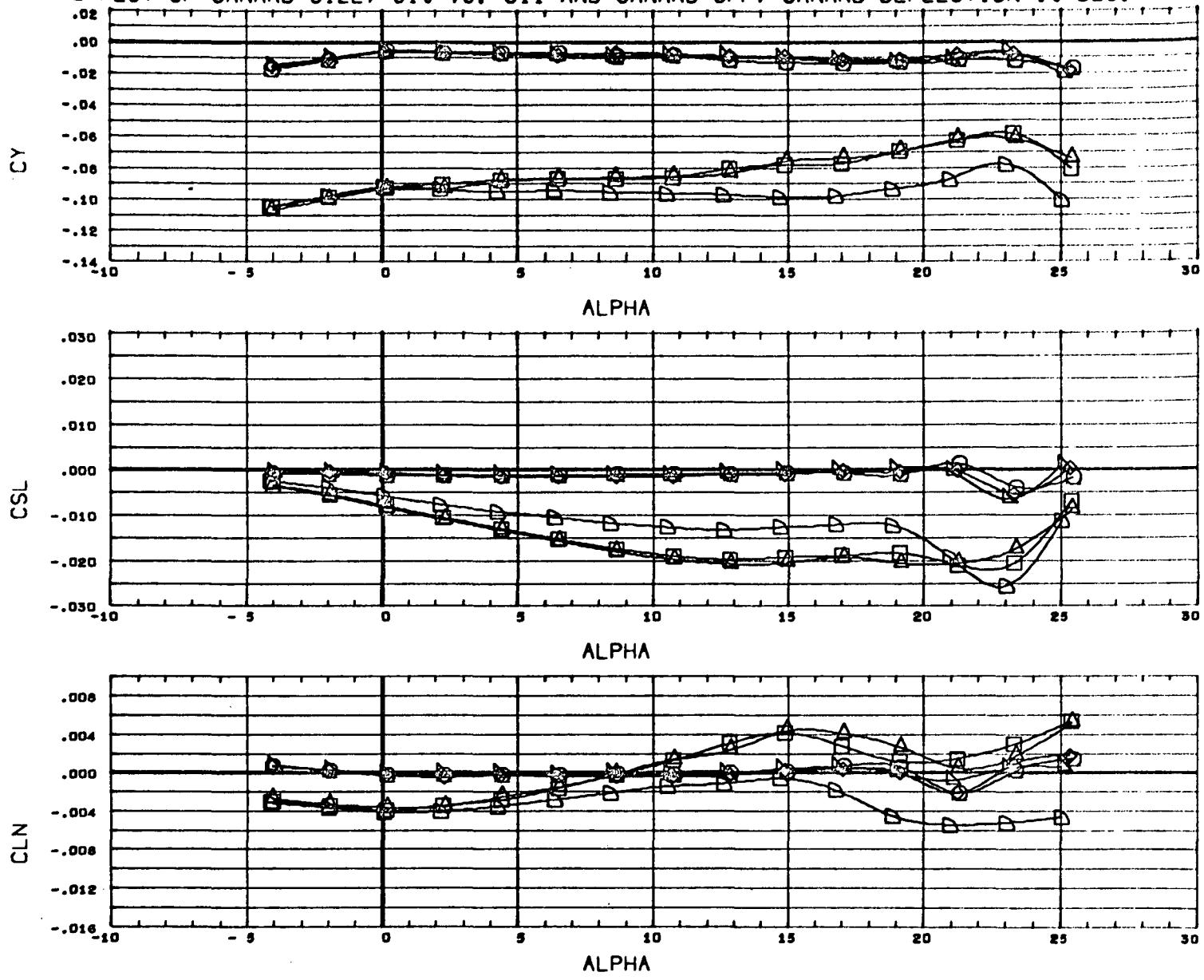
DATA SET SYMBOL	CONFIGURATION	DESCRIPTION
(AD6007)	GDLST 603-0	B30W23C10V14E38
(AD6008)	GDLST 603-0	B30W23C10V14E38
(AD6048)	GDLST 603-0	B30W23C11V14E38
(AD6049)	GDLST 603-0	B30W23C11V14E38
(AD6044)	GDLST 603-0	B30W23V14E38
(AD6045)	GDLST 603-0	B30W23V14E38

MACH 0.201



BETA	ELEVTR	RUDER	AILRON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN
5.000	0.000	0.000	0.000	LREF	16.1880 IN.
0.000	0.000	0.000	0.000	BREF	34.6320 IN.
5.000	0.000	0.000	0.000	XMRP	29.0780 IN.
0.000	0.000	0.000	0.000	YMRP	0.0000 IN.
5.000	0.000	0.000	0.000	ZMRP	0.0000 IN.
				SCALE	0.0200

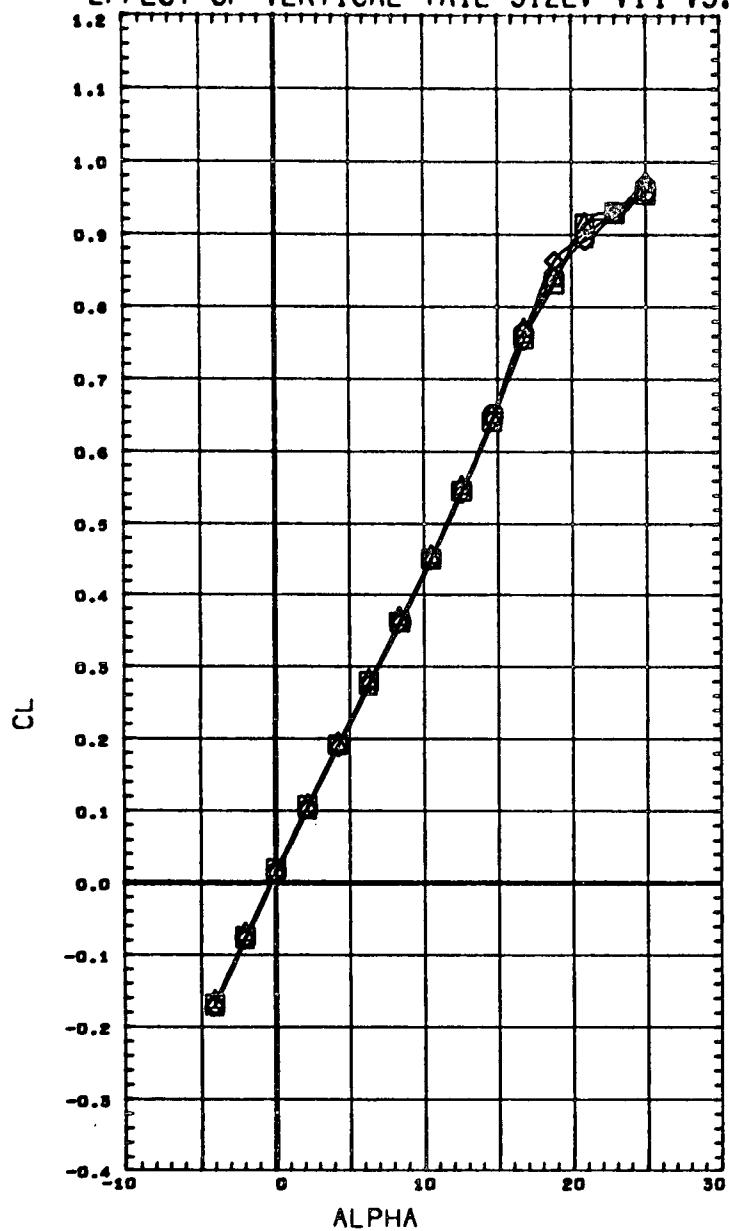
EFFECT OF CANARD SIZE, C10 VS. C11 AND CANARD OFF, CANARD DEFLECTION=10 DEG.



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	RUDER	AILRON	REFERENCE	INFORMATION
(AD6007)	GDLST 603-0 B30W23C10V14E38	0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN
(AD6008)	GDLST 603-0 B30W23C10V14E38	5.000	0.000	0.000	0.000	LREF	16.1880 IN.
(AD6048)	GDLST 603-0 B30W23C11V14E38	0.000	0.000	0.000	0.000	BREF	34.6320 IN.
(AD6049)	GDLST 603-0 B30W23C11V14E38	5.000	0.000	0.000	0.000	XMRP	29.0780 IN.
(AD6044)	GDLST 603-0 B30W23V14E38	0.000	0.000	0.000	0.000	YMRP	0.0000 IN.
(AD6045)	GDLST 603-0 B30W23V14E38	5.000	0.000	0.000	0.000	ZMRP	0.0000 IN.
						SCALE	0.0200

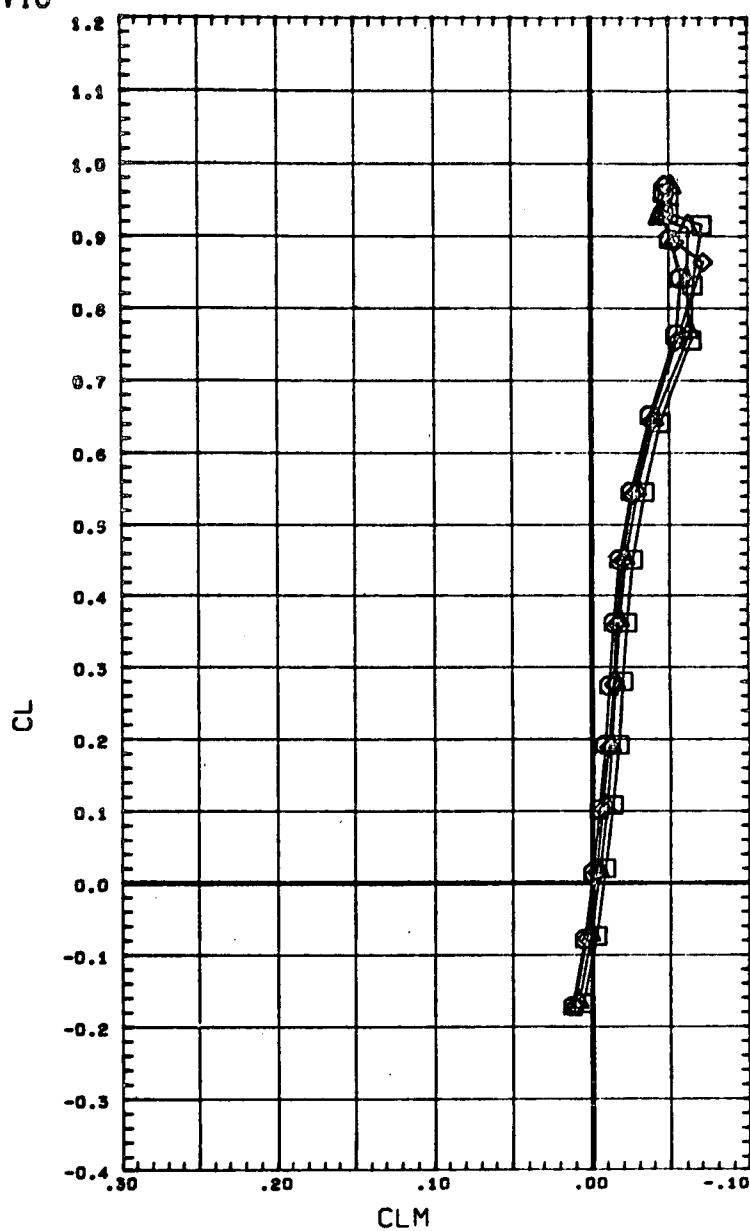
MACH 0.201

# EFFECT OF VERTICAL TAIL SIZE, V14 VS. V16



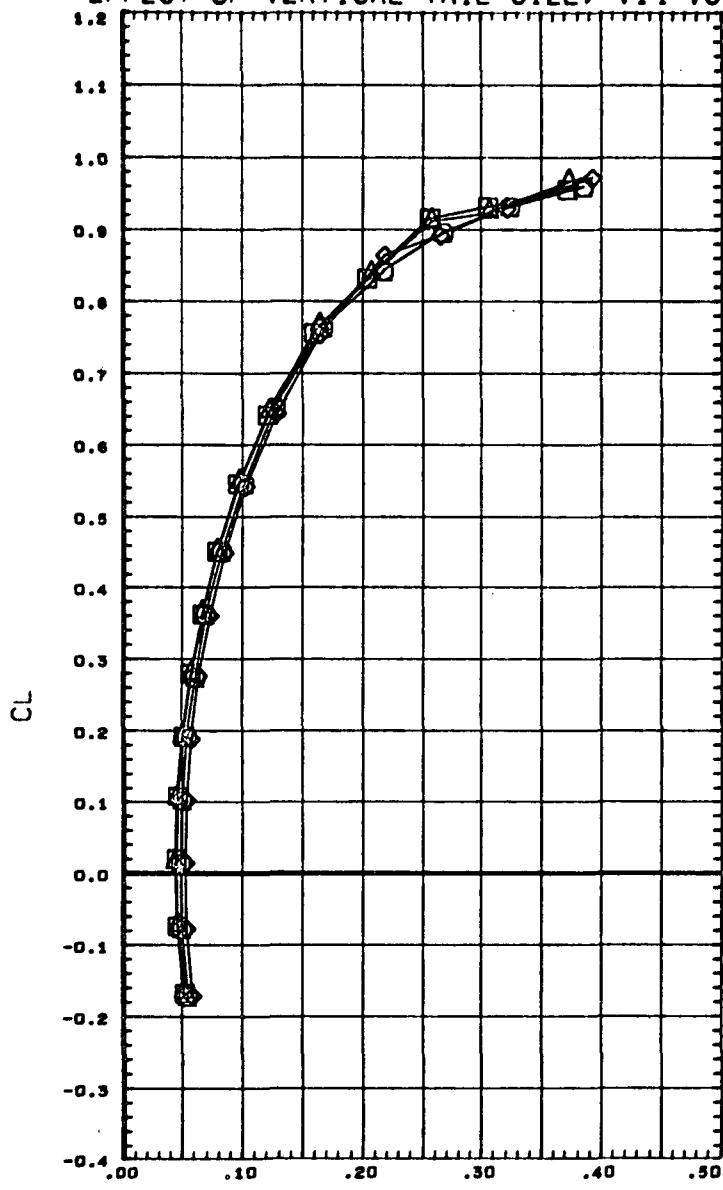
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6077) Q GDLST 603-0 830W23V14  
 (AD6078) A GDLST 603-0 830W23V14  
 (AD6079) D GDLST 603-0 830W23V16  
 (AD6080) C GDLST 603-0 830W23V16

MACH 0.201



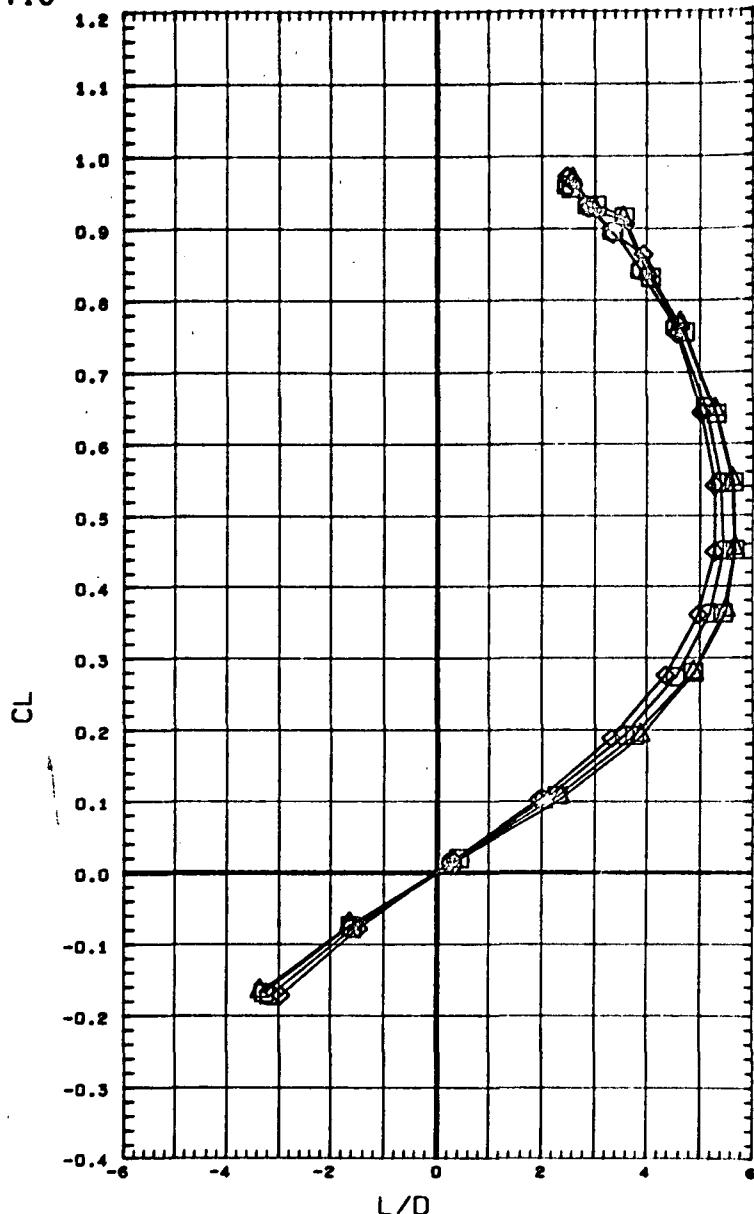
BETA	ELEVTR	RUDDER	AILRON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN.
5.000	0.000	0.000	0.000	LREF	16.1880 IN.
0.000	0.000	0.000	0.000	BREF	34.6320 IN.
5.000	0.000	0.000	0.000	XMRP	29.0760 IN.
				YMRP	0.0000 IN.
				ZMRP	0.0000 IN.
				SCALE	0.0200

# EFFECT OF VERTICAL TAIL SIZE, V14 VS. V16



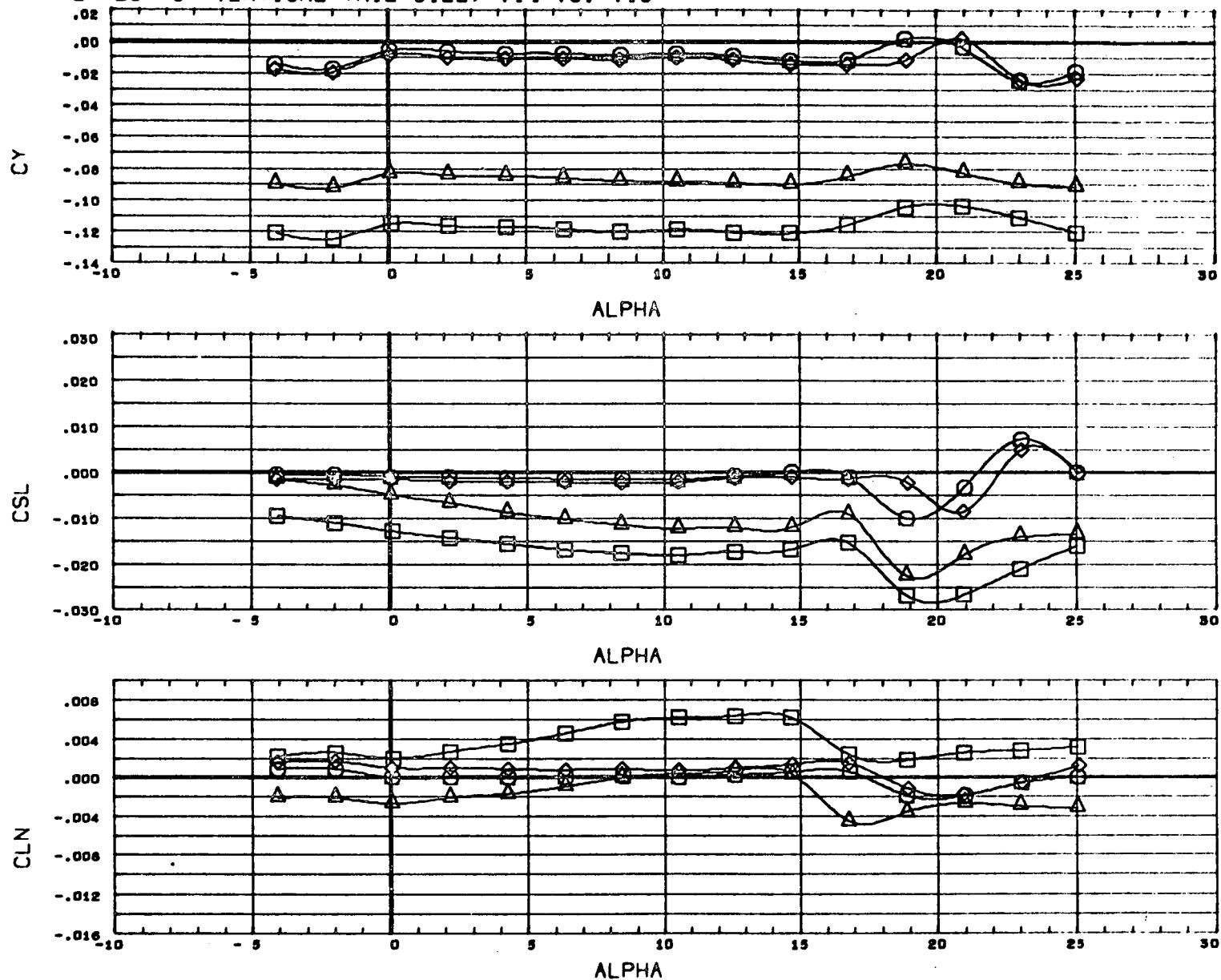
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6077) GDLST 603-0 B30W23V14  
 (AD6078) GDLST 603-0 B30W23V14  
 (AD6079) GDLST 603-0 B30W23V16  
 (AD6080) GDLST 603-0 B30W23V16

MACH 0.201



BETA	ELEVTR	RUDDER	AIRLON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN.
5.000	0.000	0.000	0.000	LREF	16.1880 IN.
0.000	0.000	0.000	0.000	BREF	34.6320 IN.
5.000	0.000	0.000	0.000	XMRP	29.0780 IN.
				YMRP	0.0000 IN.
				ZMRP	0.0000 IN.
				SCALE	0.0200

## EFFECT OF VERTICAL TAIL SIZE, V14 VS. V16



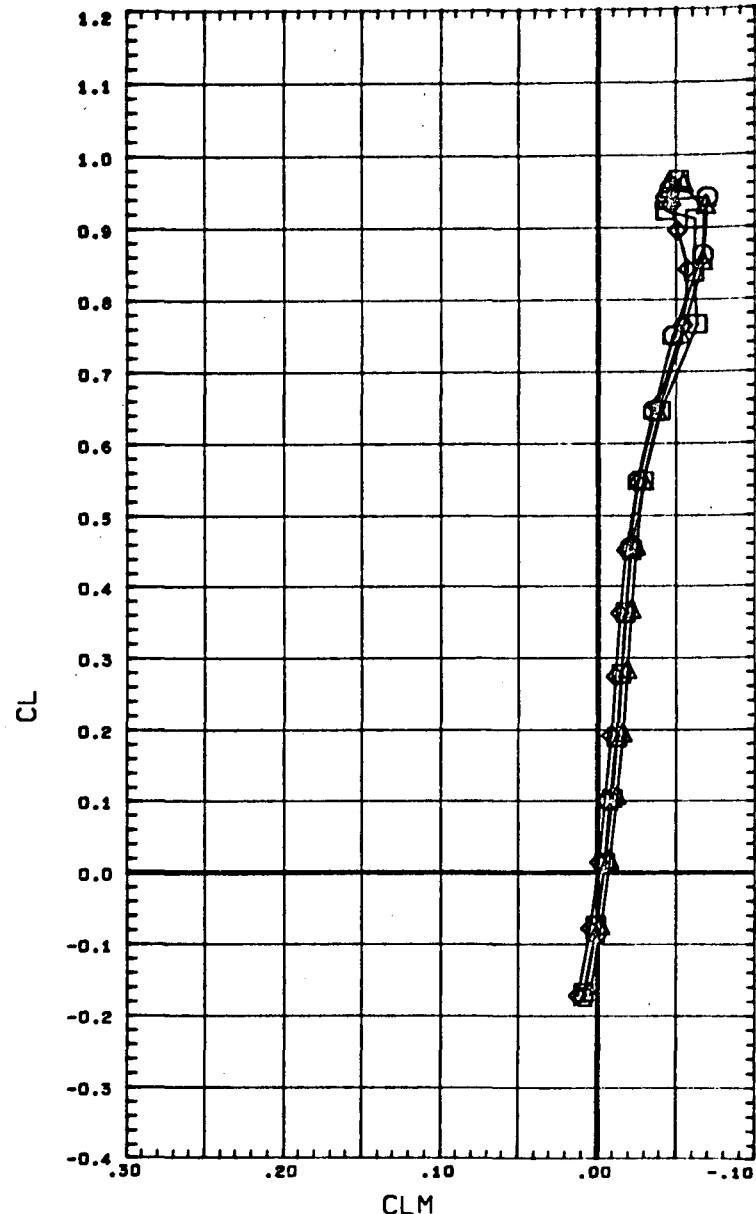
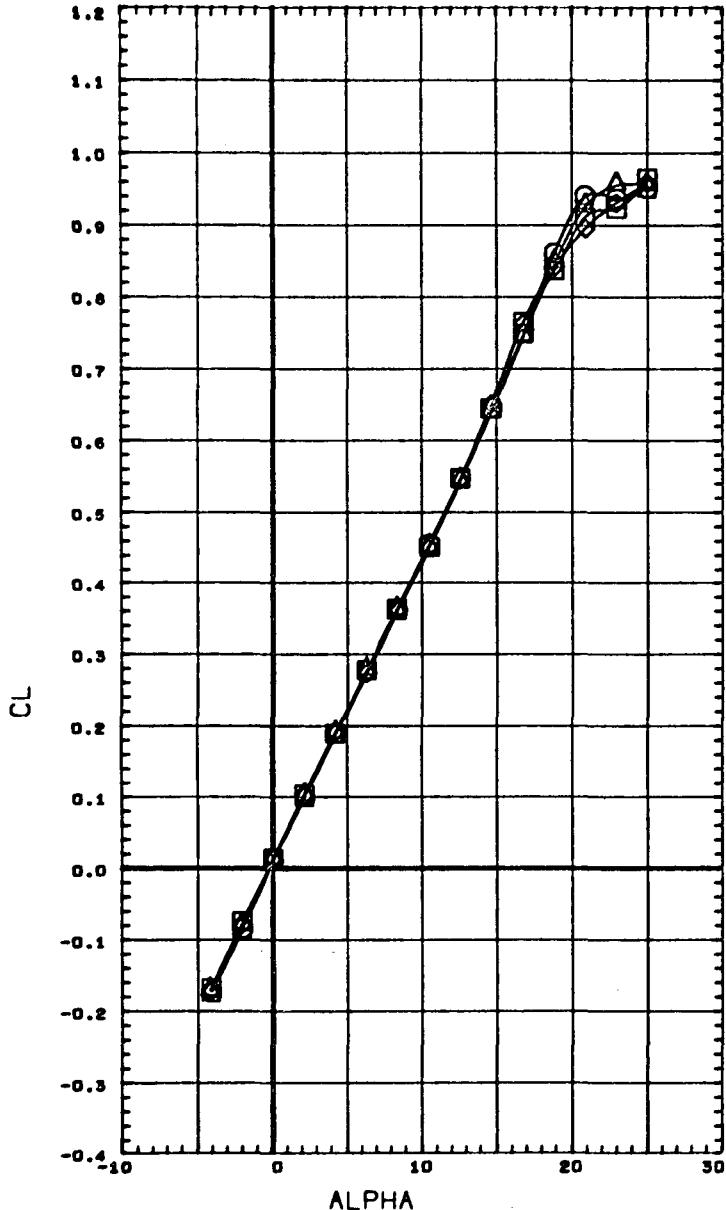
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6077) GDLST 603-0 B30W23V14  
 (AD6078) GDLST 603-0 B30W23V14  
 (AD6079) GDLST 603-0 B30W23V16  
 (AD6080) GDLST 603-0 B30W23V16

BETA	ELEVTR	RUDER	AIRLON	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN.
5.000	0.000	0.000	0.000	LREF 16.1880 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
5.000	0.000	0.000	0.000	XMRP 29.0780 IN.
				YMRP 0.0000 IN.
				ZMRP 0.0000 IN.
				SCALE 0.0200

MACH 0.201

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## EFFECT OF FLOW-THRU ENGINES E38

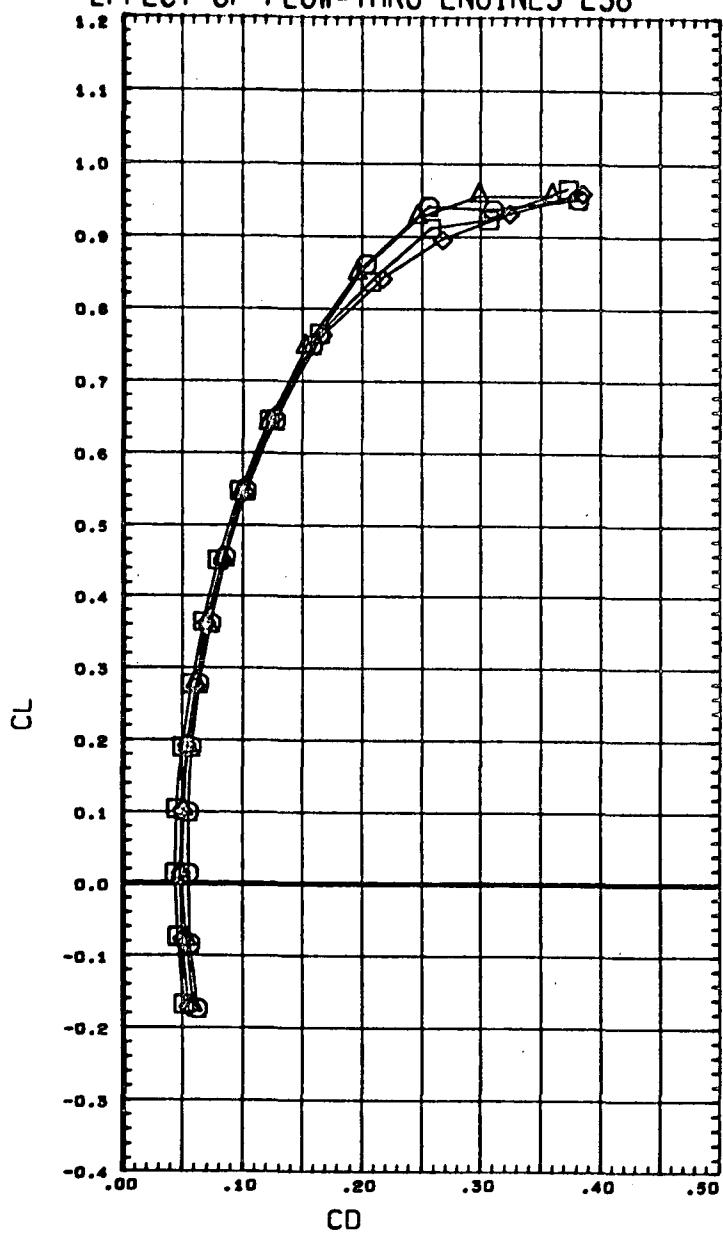


DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6044) GDLST 603-0 830W23V14E38  
 (AD6045) GDLST 603-0 830W23V14E38  
 (AD6077) GDLST 603-0 830W23V14  
 (AD6078) GDLST 603-0 830W23V14

MACH 0.201

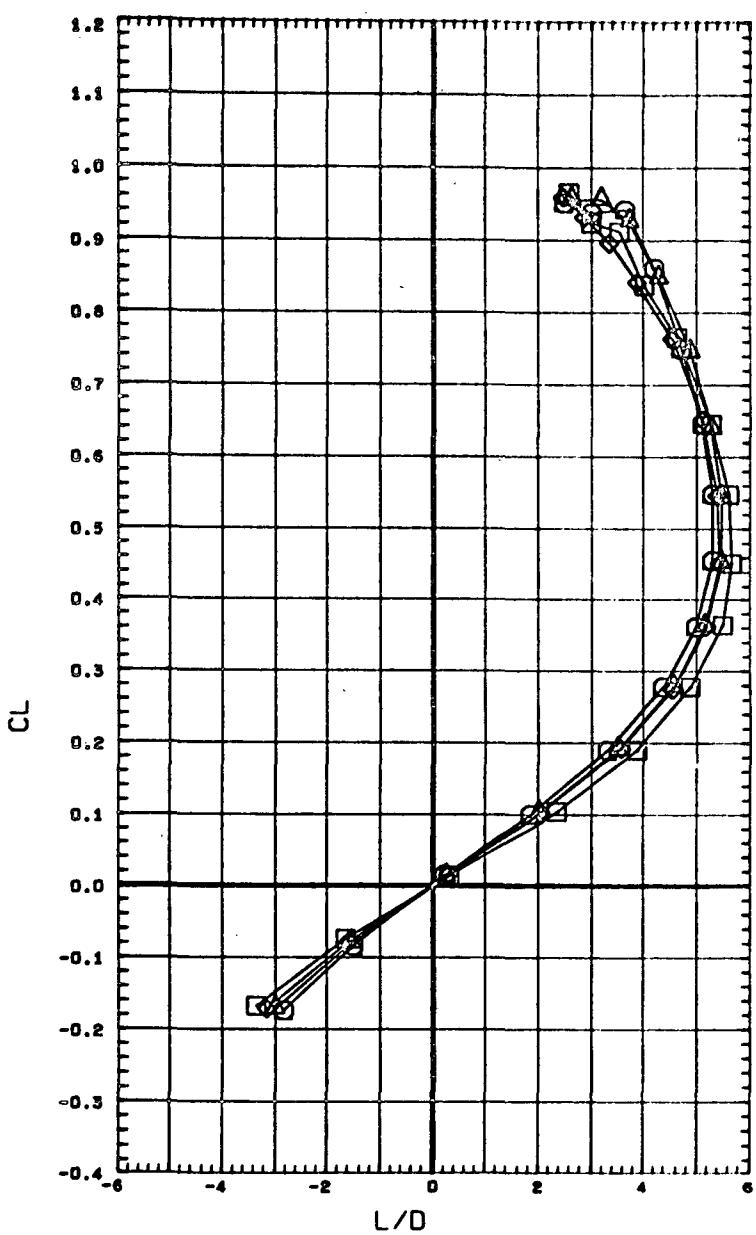
BETA	ELEVTR	RUDER	AIRLON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN.
5.000	0.000	0.000	0.000	LREF	16.1880 IN.
0.000	0.000	0.000	0.000	BREF	34.6320 IN.
5.000	0.000	0.000	0.000	XMRP	29.0780 IN.
				YMRP	0.0000 IN.
				ZMRP	0.0000 IN.
				SCALE	0.0200

## EFFECT OF FLOW-THRU ENGINES E38



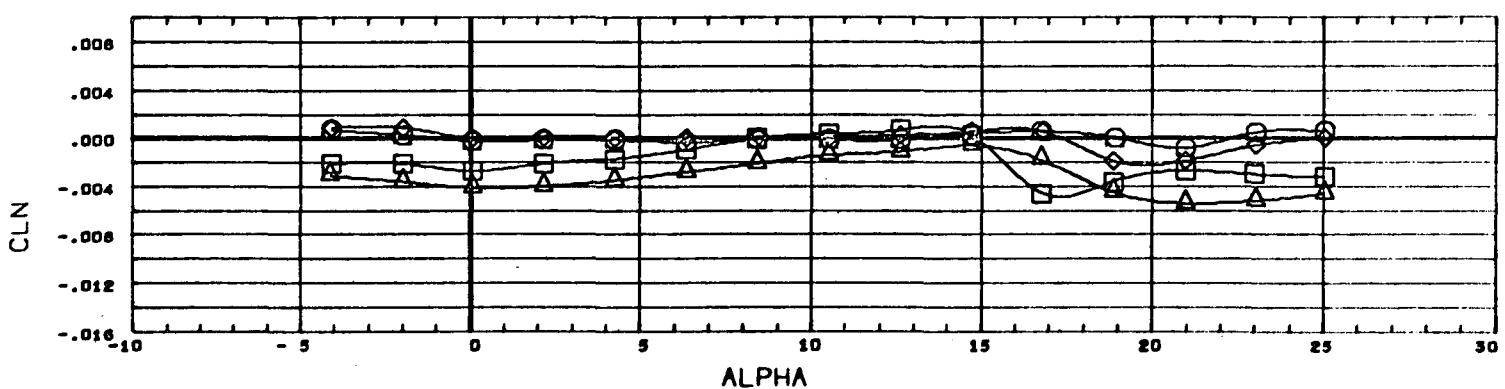
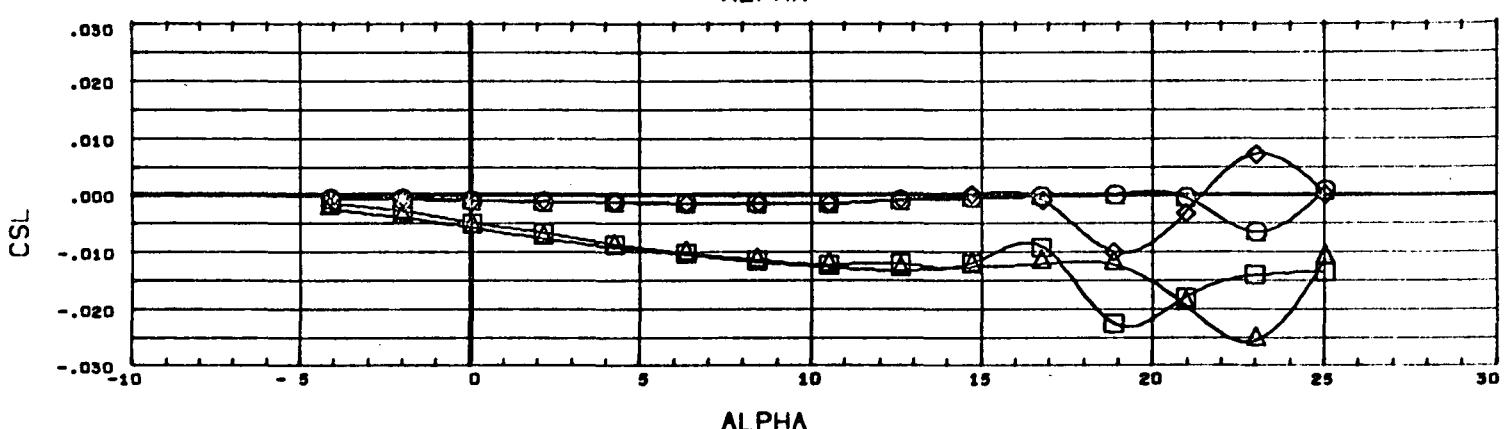
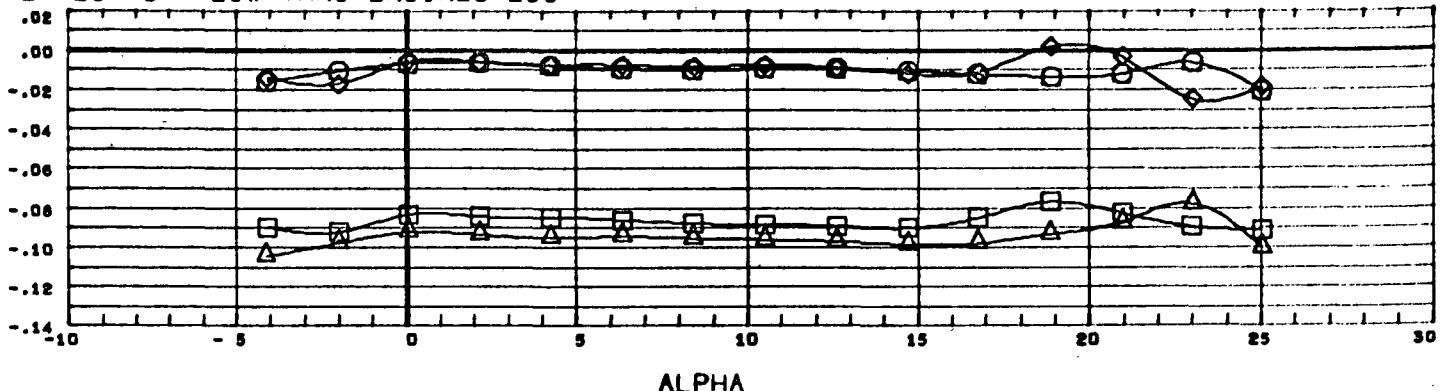
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AD044)	GDLST 603-0 B30W23V14E38
(AD045)	GDLST 603-0 B30W23V14E38
(AD077)	GDLST 603-0 B30W23V14
(AD078)	GDLST 603-0 B30W23V14

MACH 0.201



BETA	ELEVTR	RUDDER	AIRLON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN
5.000	0.000	0.000	0.000	LREF	16.1880 IN.
0.000	0.000	0.000	0.000	BREF	34.6320 IN.
5.000	0.000	0.000	0.000	XHRP	29.0760 IN.
				YHRP	0.0000 IN.
				ZHRP	0.0000 IN.
				SCALE	0.0200

## EFFECT OF FLOW-THRU ENGINES E38



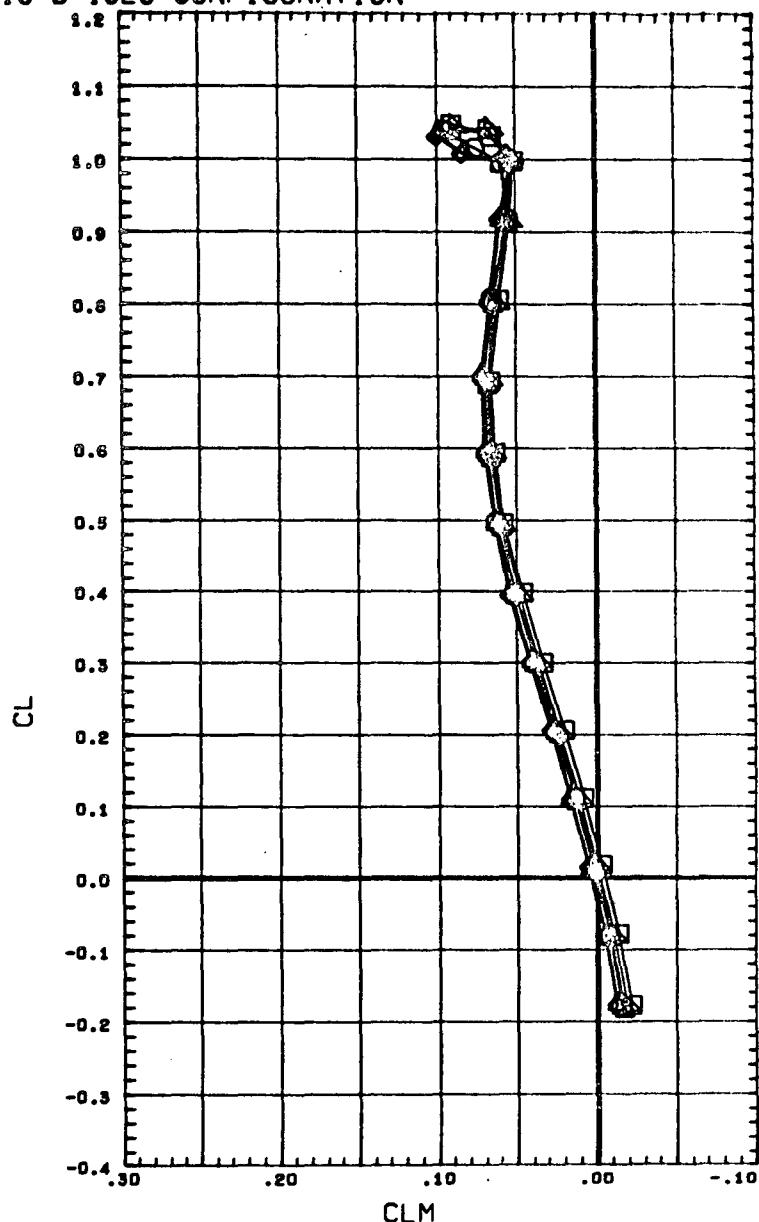
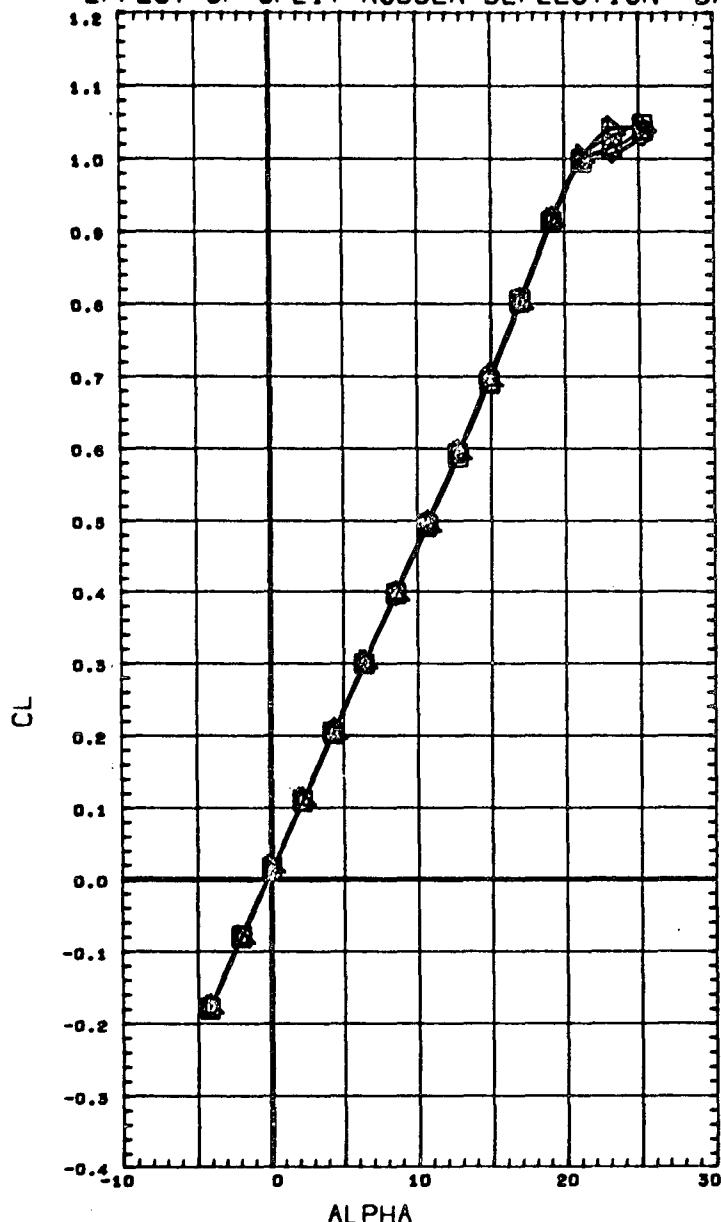
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AD6044)	GDLST 603-0 B30W23V14E38
(AD6045)	GDLST 603-0 B30W23V14E38
(AD6077)	GDLST 603-0 B30W23V14
(AD6078)	GDLST 603-0 B30W23V14

BETA	ELEVTR	RUDDER	AIRLON	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN.
5.000	0.000	0.000	0.000	LREF 16.1680 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
5.000	0.000	0.000	0.000	XMRP 29.0780 IN.
				YMRP 0.0000 IN.
				ZMRP 0.0000 IN.
				SCALE 0.0200

MACH 0.201

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# EFFECT OF SPLIT RUDDER DEFLECTION- BASIC B-18E3 CONFIGURATION

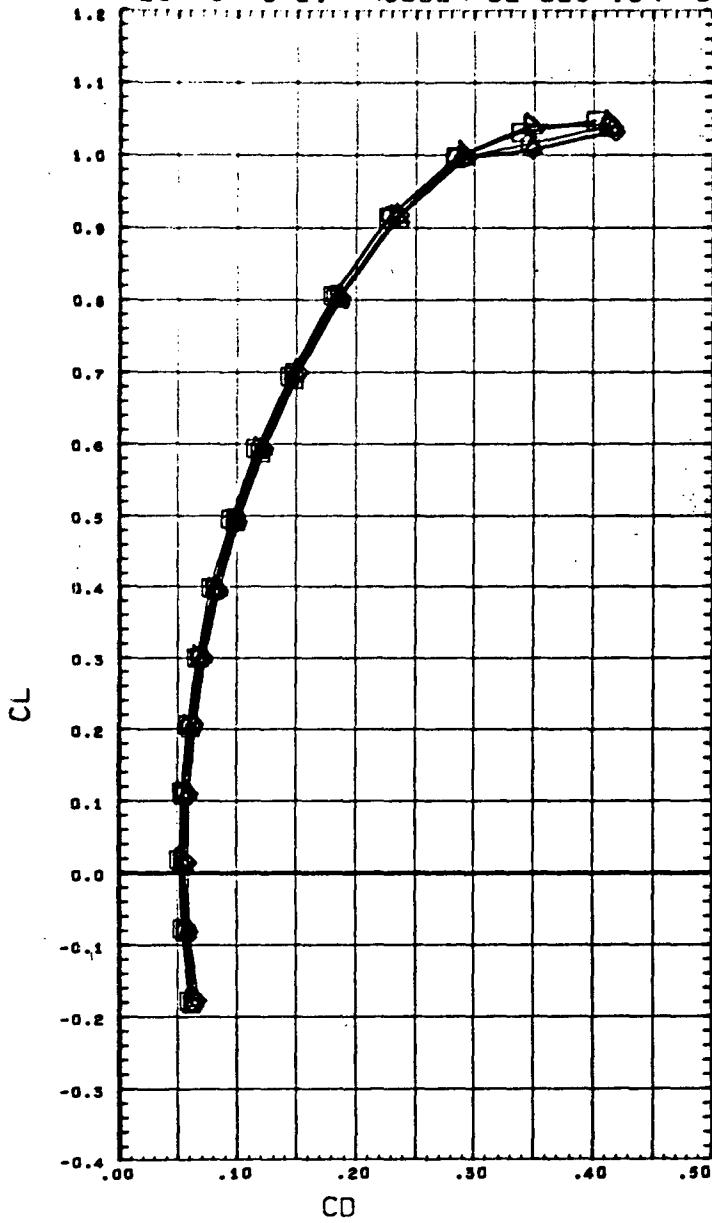


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AD6002)	GDLST 603-0 B30W23C10V14E38
(AD6040)	GDLST 603-0 B30W23C10V14E38 (SPLIT RUDDER)
(AD6036)	GDLST 603-0 B30W23C10V14E38 (RUDDER)
(AD6003)	GDLST 603-0 B30W23C10V14E38
(AD6041)	GDLST 603-0 B30W23C10V14E38 (SPLIT RUDDER)
(AD6039)	GDLST 603-0 B30W23C10V14E38 (RUDDER)

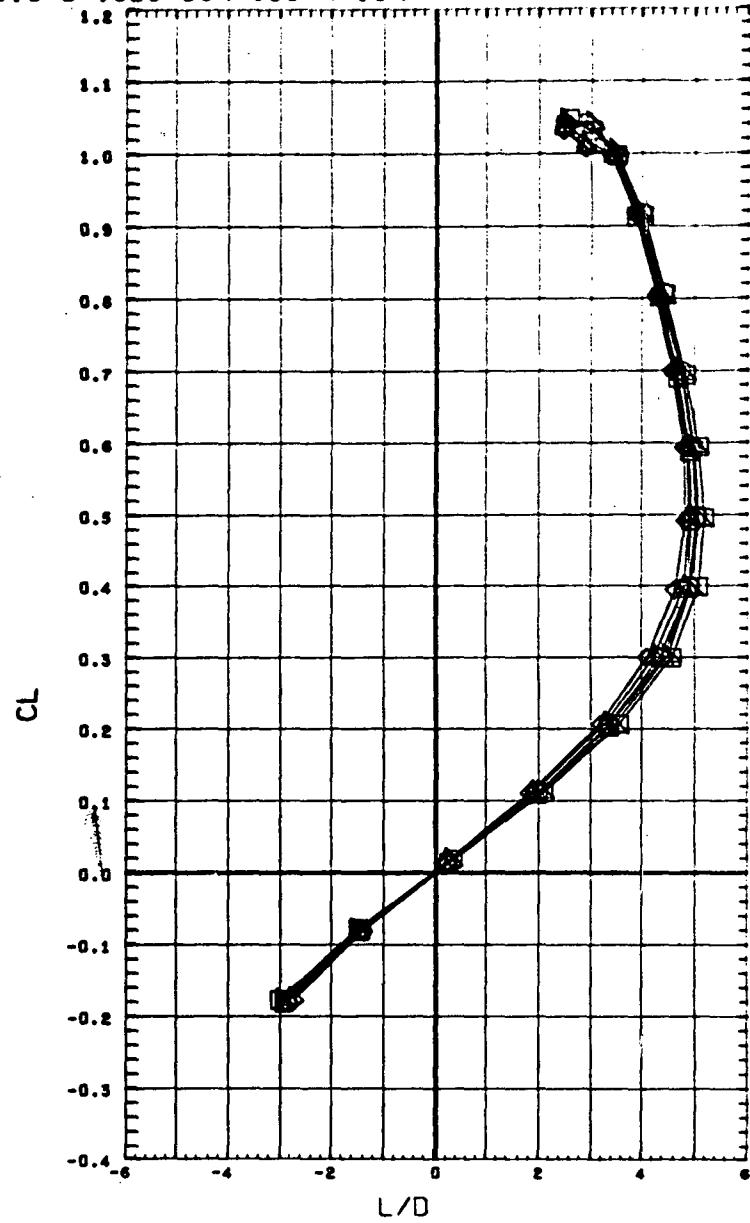
MACH 0.201

BETA	ELEVTR	CANARD	RUDDER	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN
0.000	0.000	0.000	5.000	LREF	16.1880 IN.
0.000	0.000	0.000	10.000	BREF	34.6320 IN.
5.000	0.000	0.000	0.000	XMRP	29.0780 IN.
5.000	0.000	0.000	5.000	YMRP	0.0000 IN.
5.000	0.000	0.000	10.000	ZMRP	0.0000 IN.
				SCALE	0.0200

## EFFECT OF SPLIT RUDDER DEFLECTION- BASIC B-18E3 CONFIGURATION

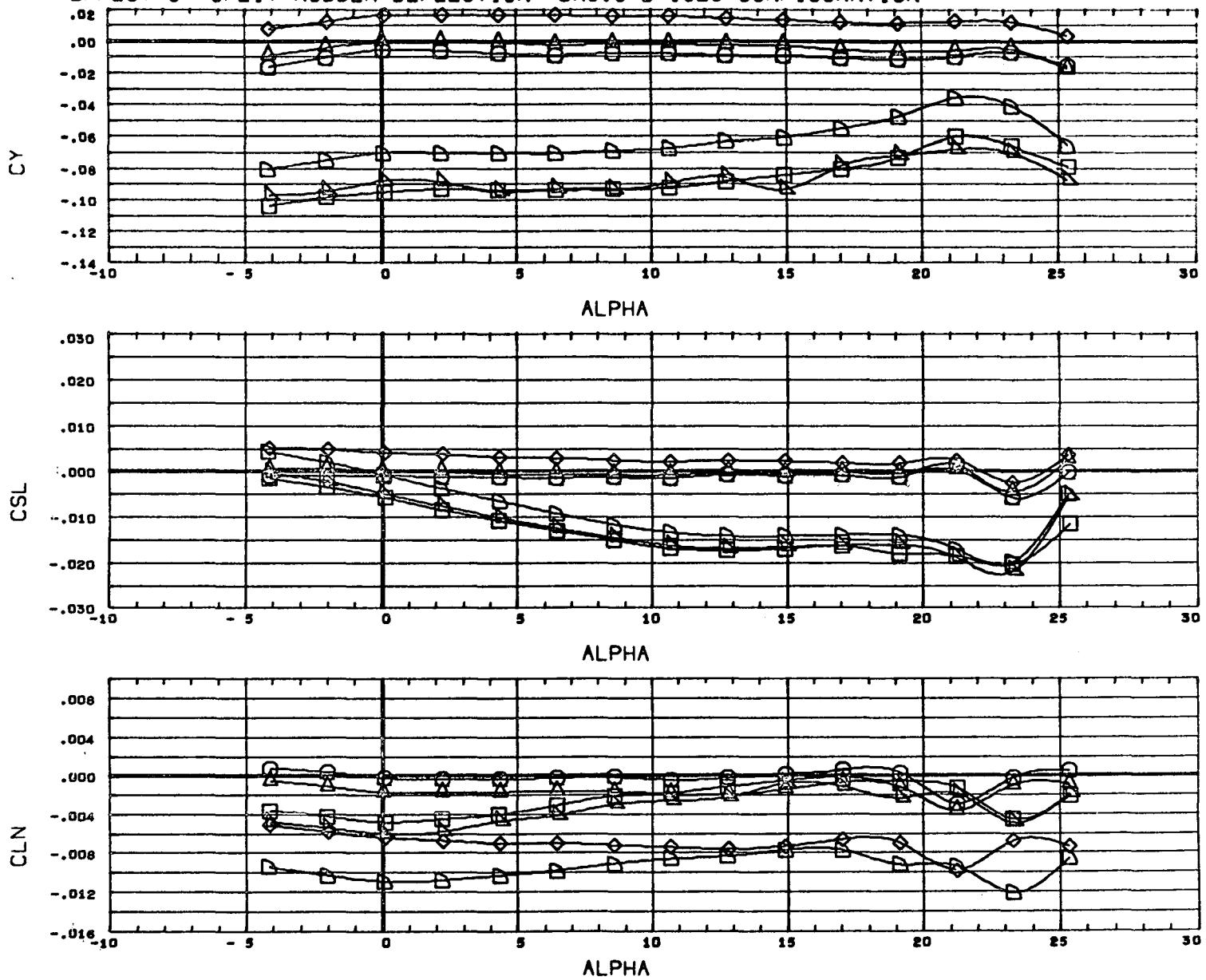


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AC6002)	GCLST 603-0 B30W23C10V14E38
(AC6049)	GCLST 603-0 B30W23C10V14E38 (SPLIT RUDDER)
(AC6038)	GCLST 603-0 B30W23C10V14E38 (RUDDER)
(AC6003)	GCLST 603-0 B30W23C10V14E38
(AC6041)	GCLST 603-0 B30W23C10V14E38 (SPLIT RUDDER)
(AC6039)	GCLST 603-0 B30W23C10V14E38 (RUDDER)



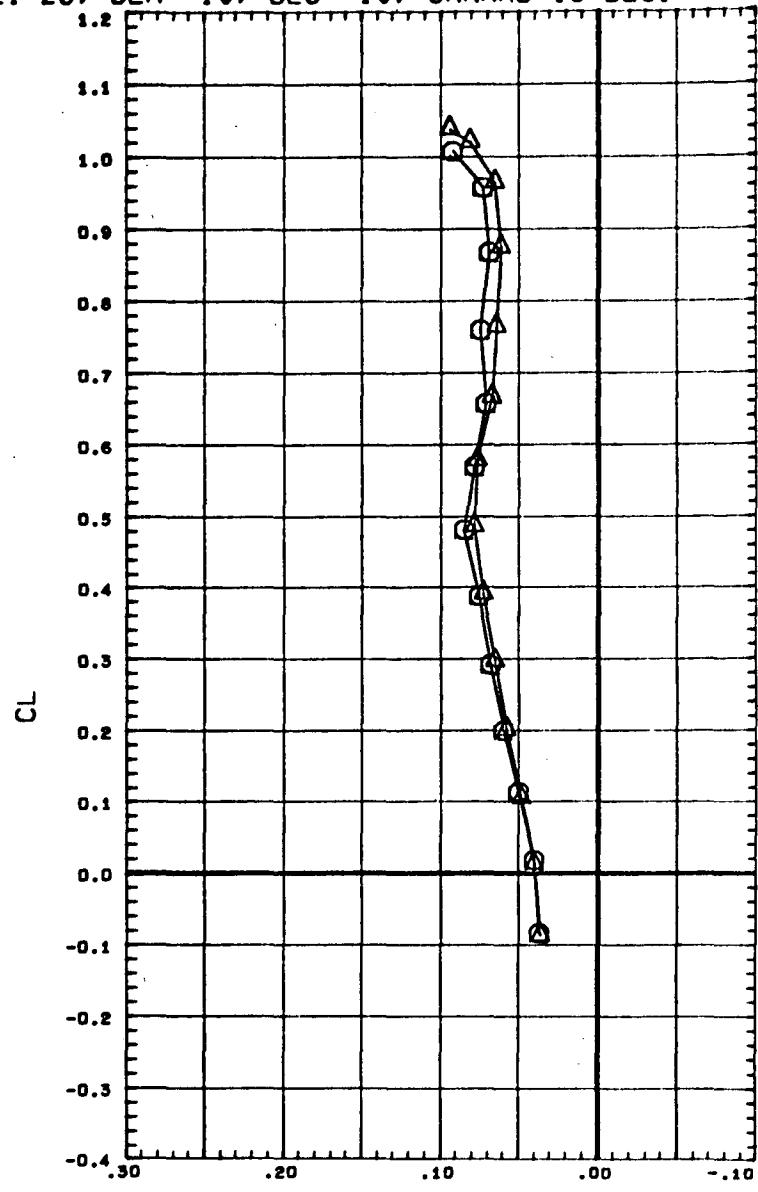
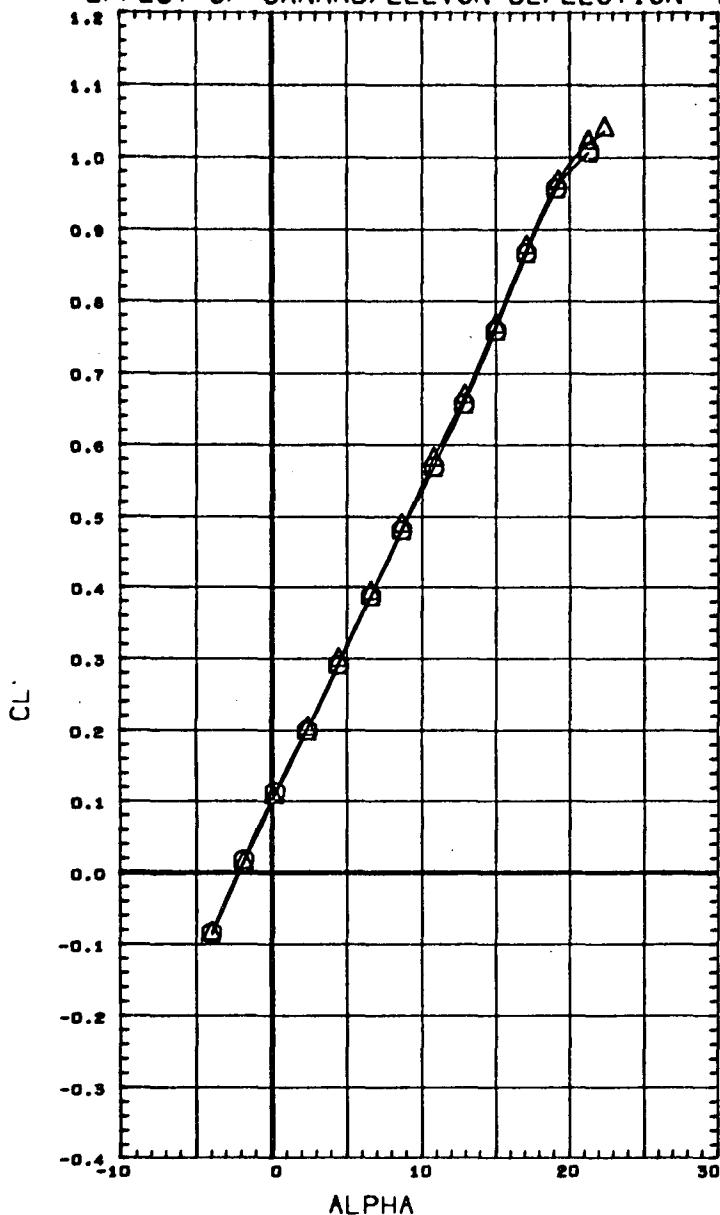
BETA	ELEVTR	CANARC	RUCDER	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN
0.000	0.000	5.000	5.000	LREF 16.1880 IN.
0.000	0.000	0.000	10.000	BREF 34.6320 IN.
5.000	0.000	0.000	0.000	XMRP 29.0780 IN.
5.000	0.000	0.000	5.000	YMRP 0.0000 IN.
5.000	0.000	0.000	10.000	ZMRP 0.0000 IN.
				SCALE 0.0200

### EFFECT OF SPLIT RUDDER DEFLECTION- BASIC B-18E3 CONFIGURATION



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	RUDDER	AILRON	REFERENCE INFORMATION
(AD6002)	GDLST 603-0 B30W23C10V14E38	0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN.
(AD6040)	GDLST 603-0 B30W23C10V14E38 (SPLIT RUDDER)	0.000	0.000	5.000	0.000	LREF 16.1680 IN.
(AD6038)	GDLST 603-0 B30W23C10V14E38 (RUDDER)	0.000	0.000	10.000	0.000	BREF 34.6320 IN.
(AD6003)	GDLST 603-0 B30W23C10V14E38	5.000	0.000	0.000	0.000	XMRP 29.0780 IN.
(AD6041)	GDLST 603-0 B30W23C10V14E38 (SPLIT RUDDER)	5.000	0.000	5.000	0.000	YMRP 0.0000 IN.
(AD6039)	GDLST 603-0 B30W23C10V14E38 (RUDDER)	5.000	0.000	10.000	0.000	ZMRP 0.0000 IN.
MACH	0.201					SCALE 0.0200

EFFECT OF CANARD/ELEVON DEFLECTION- DEI=25, DEM=-10, DEO=-10, CANARD=15 DEG.

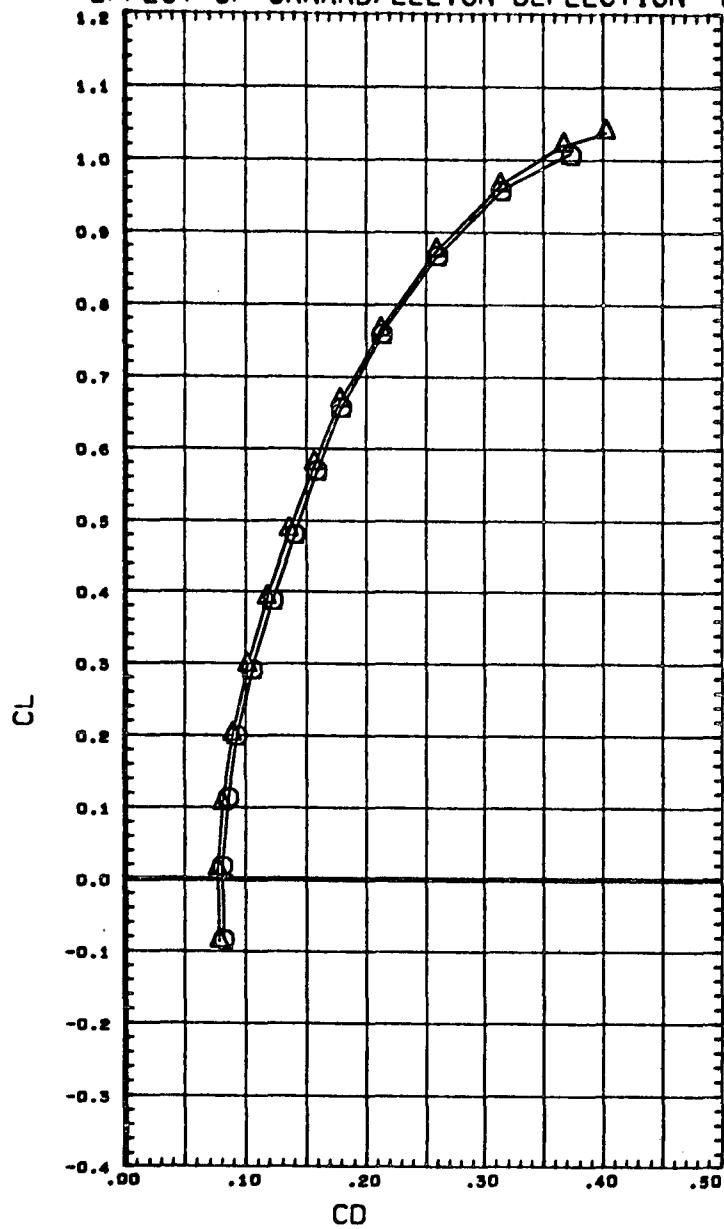


DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6071) Q GDLST 603-0 B30W23C10V14E38 (CANARD + ELEVON)  
 (AD6072) Q GDLST 603-0 B30W23C10V14E38 (CANARD + ELEVON)

MACH 0.201

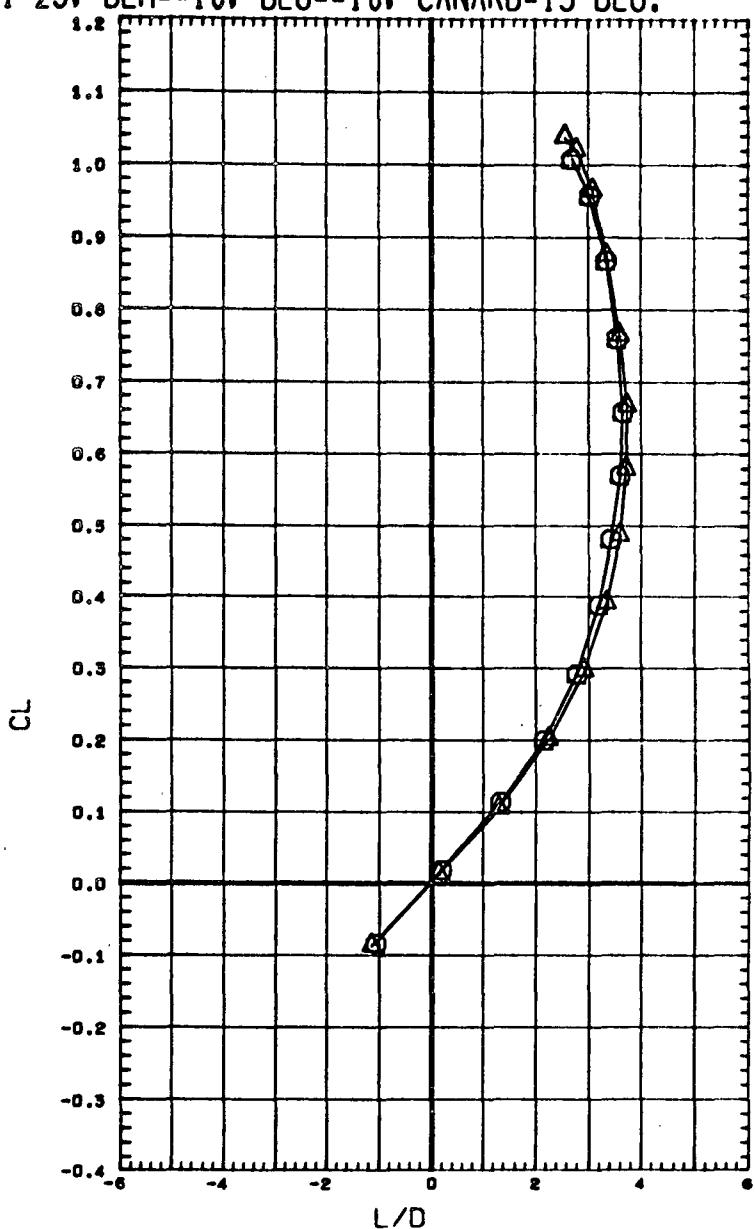
BETA	ELEVTR	RUDER	AILRON	REFERENCE	INFORMATION
0.000	1.660	0.000	0.000	SREF	492.4804 SQ. IN.
5.000	1.660	0.000	0.000	LREF	16.1880 IN.
				BREF	34.6320 IN.
				XMRP	29.0780 IN.
				YMRP	0.0000 IN.
				ZMRP	0.0000 IN.
				SCALE	0.0200

EFFECT OF CANARD/ELEVON DEFLECTION- DEI=25, DEM=-10, DEO=-10, CANARD=15 DEG.



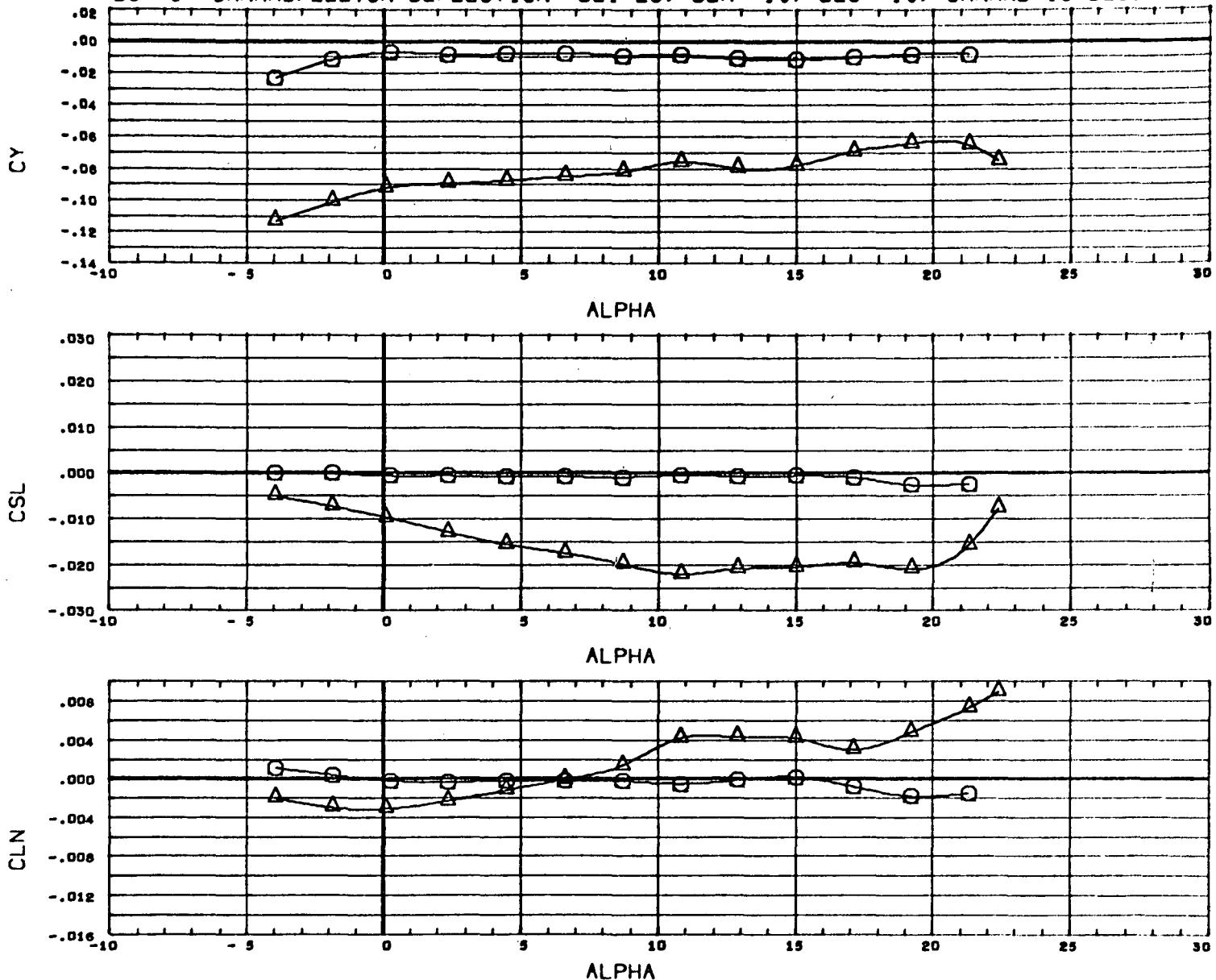
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6071) GDLST 603-0 830W23C10V14E38(CANARD + ELEVON)  
 (AD6072) GDLST 603-0 830W23C10V14E38(CANARD + ELEVON)

MACH 0.201



BETA	ELEVTR	RUDER	AIRLON	REFERENCE INFORMATION
0.000	1.660	0.000	0.000	SREF 492.4804 SQ. IN.
5.000	1.660	0.000	0.000	LREF 16.1680 IN.
				BREF 34.6320 IN.
				XMRP 29.0760 IN.
				YMRP 0.0000 IN.
				ZMRP 0.0000 IN.
				SCALE 0.0200

EFFECT OF CANARD/ELEVON DEFLECTION- DEI=25, DEM=-10, DEO=-10, CANARD=15 DEG.

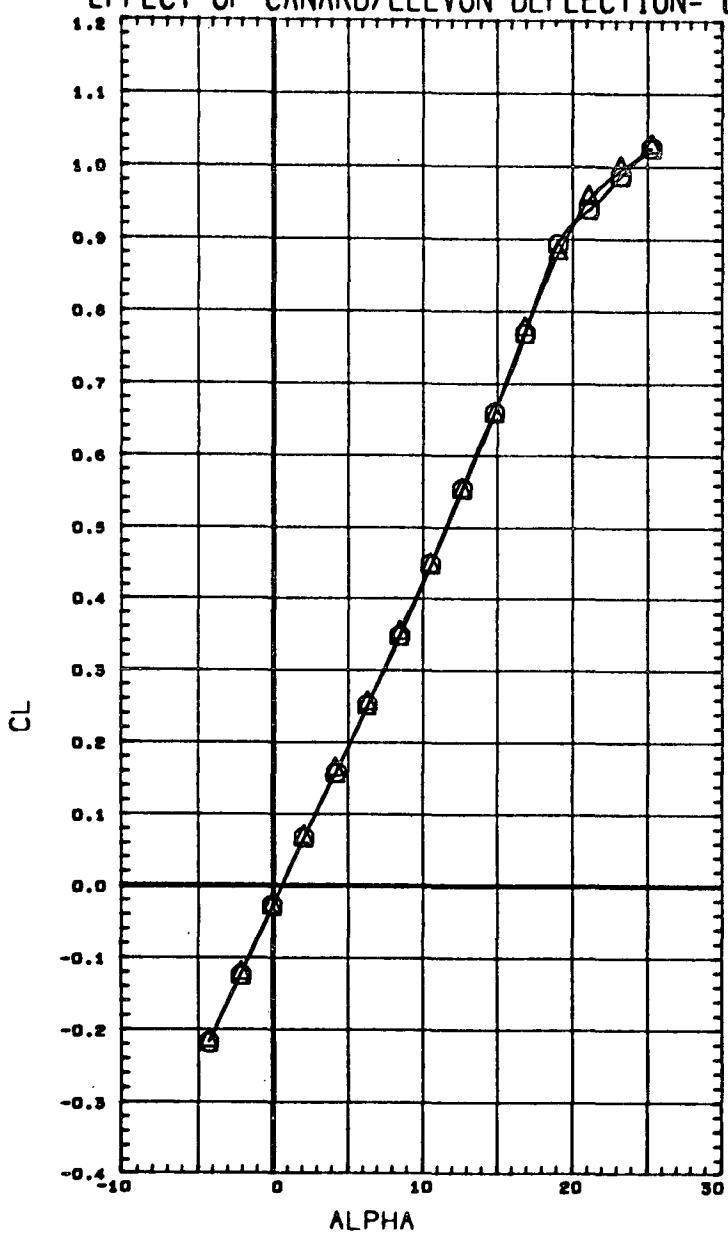


DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	RUDDER	AIRLON	REFERENCE INFORMATION
(AD6071)	GDLST 603-D B30W23C10V14E38 (CANARD + ELEVON)	0.000	1.660	0.000	0.000	SREF 492.4804 SQ. IN
(AD6072)	GDLST 603-D B30W23C10V14E38 (CANARD + ELEVON)	5.000	1.660	0.000	0.000	LREF 16.1680 IN.
					BREF 34.6320 IN.	
					XMRP 29.0780 IN.	
					YMRP 0.0000 IN.	
					ZMRP 0.0000 IN.	
					SCALE 0.0200	

MACH 0.201

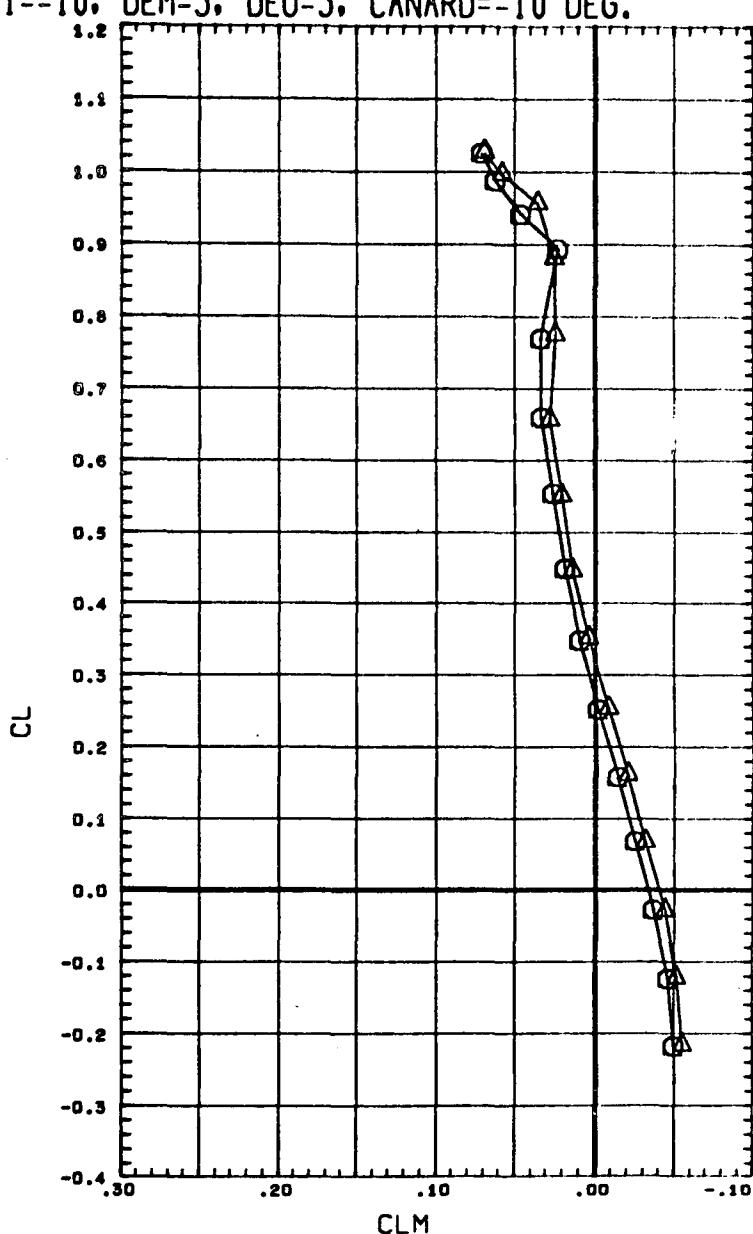
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EFFECT OF CANARD/ELEVON DEFLECTION- DEI=-10, DEM=5, DE0=5, CANARD=-10 DEG.



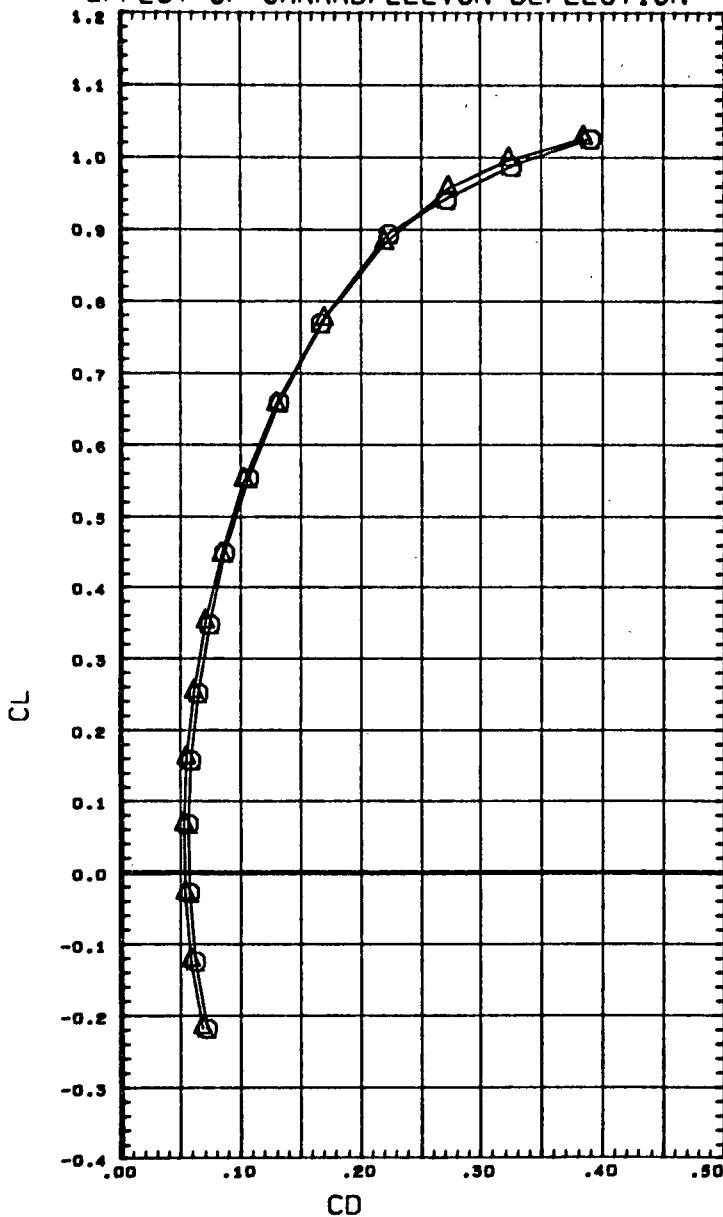
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6073) O GDLST 603-0 B30W23C10V14E38 (CANARD + ELEVON)  
 (AD6074) Q GDLST 603-0 B30W23C10V14E38 (CANARD + ELEVON)

MACH 0.201



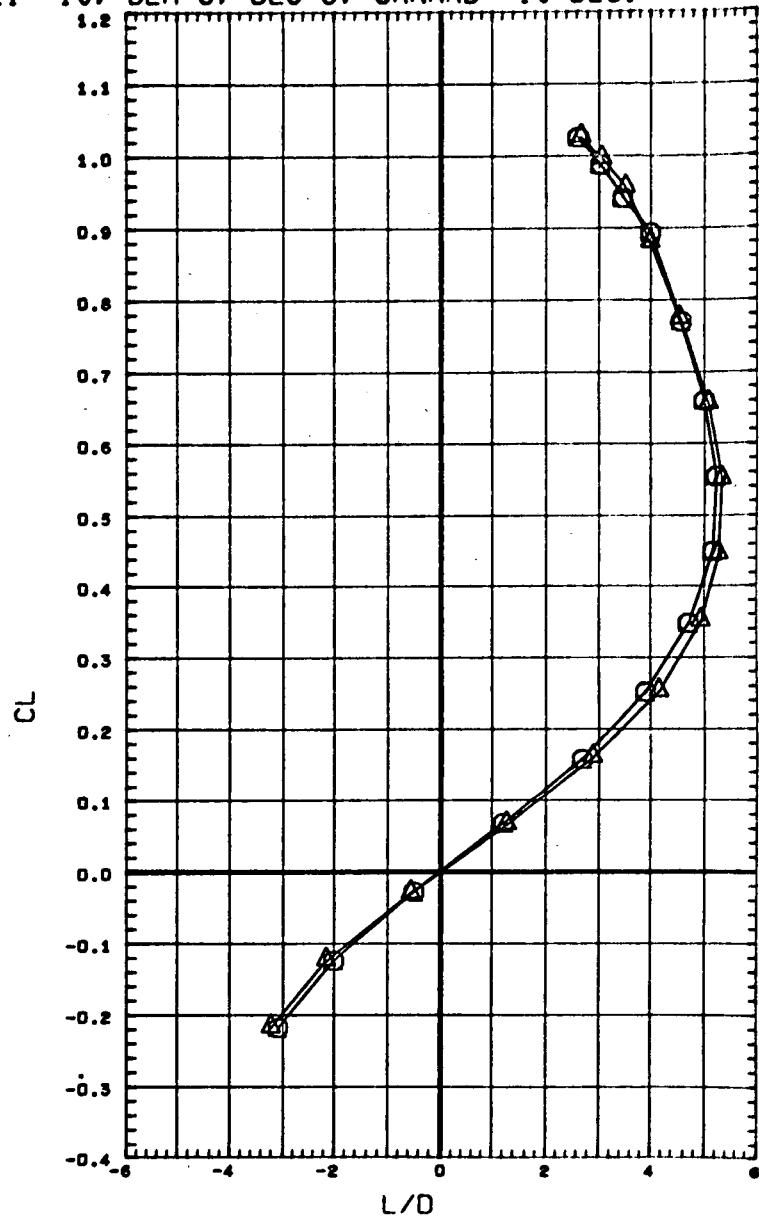
BETA ELEVTR RUDDER AILRON REFERENCE INFORMATION  
 0.000 0.000 0.000 0.000 SREF 492.4804 SQ. IN.  
 5.000 0.000 0.000 0.000 LREF 16.1880 IN.  
 BREF 34.6320 IN.  
 XMRP 29.0780 IN.  
 YMRP 0.0000 IN.  
 ZMRP 0.0000 IN.  
 SCALE 0.0200

EFFECT OF CANARD/ELEVON DEFLECTION- DEI=-10, DEM=5, DE0=5, CANARD=-10 DEG.



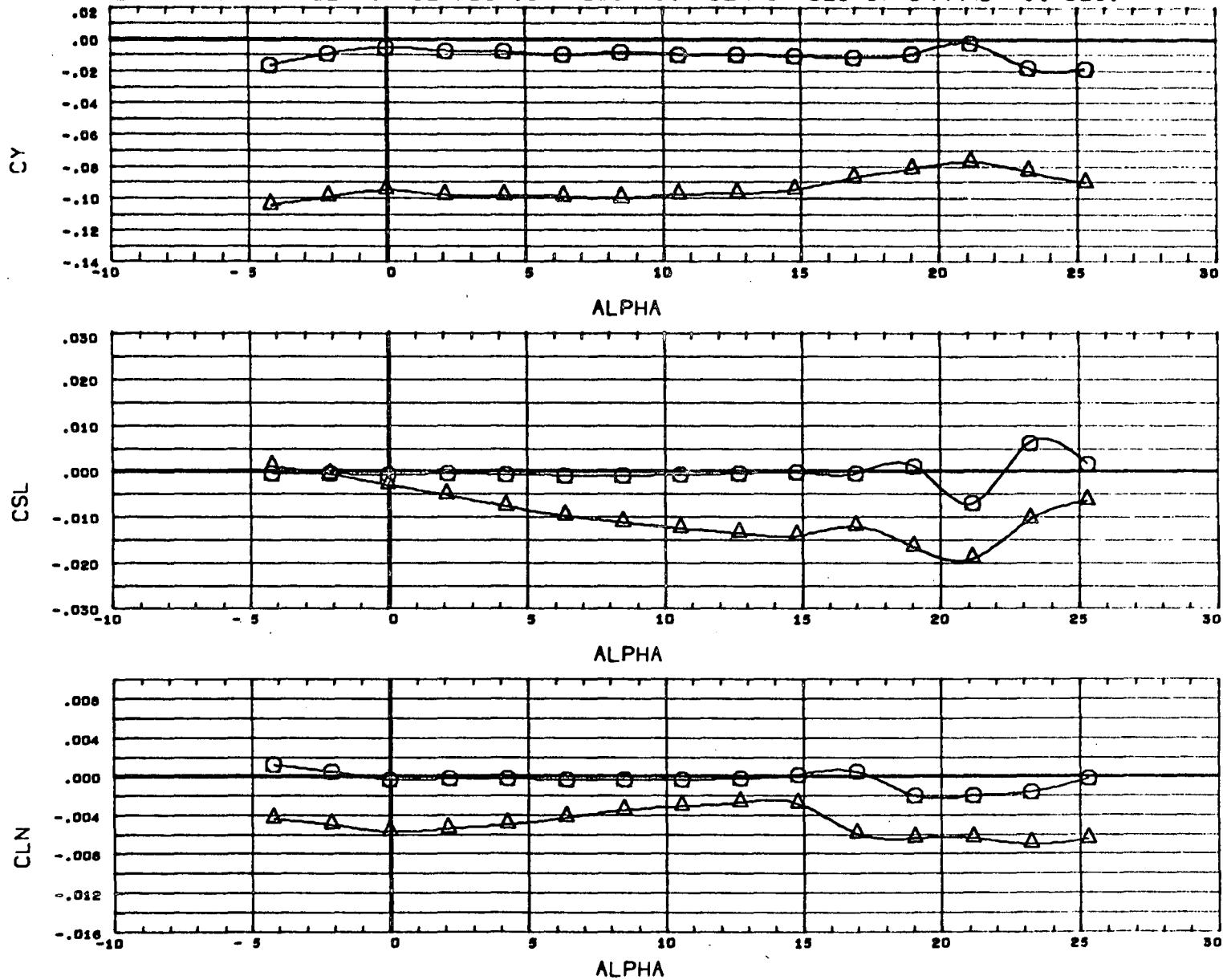
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6073) Q GDLST 603-0 B30W23C10V14E38 (CANARD + ELEVON)  
 (AD6074) A GDLST 603-0 B30W23C10V14E38 (CANARD + ELEVON)

MACH 0.201



BETA	ELEVTR	RUDER	AIRLN	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN.
5.000	0.000	0.000	0.000	LREF 16.1880 IN.
				BREF 34.6320 IN.
				XMRP 29.0780 IN.
				YMRP 0.0000 IN.
				ZMRP 0.0000 IN.
				SCALE 0.0200

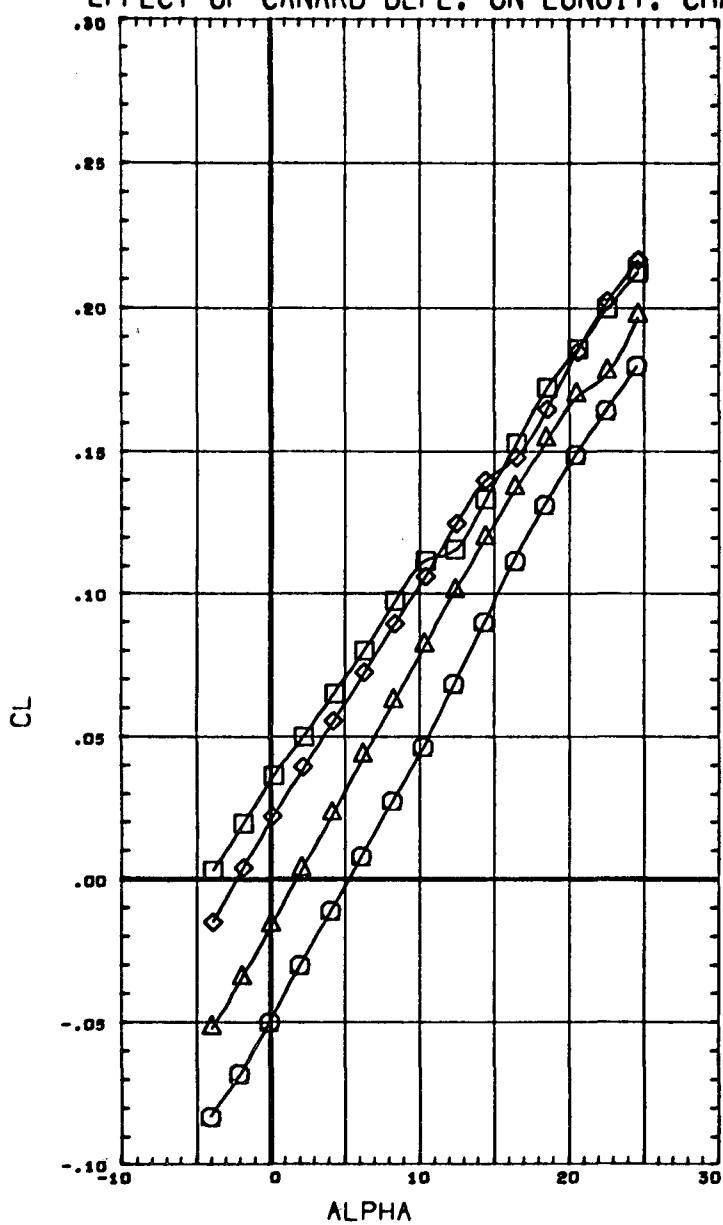
EFFECT OF CANARD/ELEVON DEFLECTION- DEI=-10, DEM=5, DE0=5, CANARD=-10 DEG.



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	RUDDER	AIRLON	REFERENCE INFORMATION
(AD6073)	GDLST 603-0 B30W23C10V14E38 (CANARD + ELEVON)	0.000	0.000	0.000	0.000	SREF 492.4604 SQ. IN
(AD6074)	GDLST 603-0 B30W23C10V14E38 (CANARD + ELEVON)	5.000	0.000	0.000	0.000	LREF 16.1880 IN.
					BREF 34.6320 IN.	
					XMRP 29.0780 IN.	
					YMRP 0.0000 IN.	
					ZMRP 0.0000 IN.	
					SCALE 0.0200	

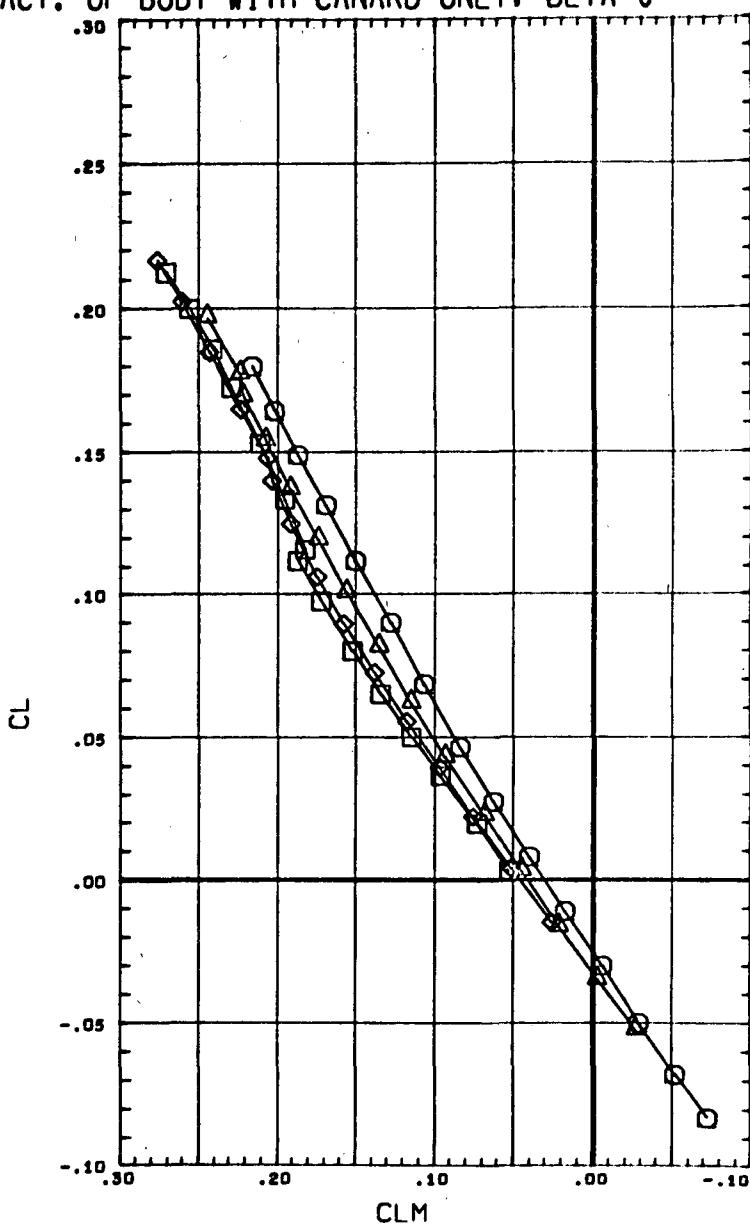
MACH 0.201

EFFECT OF CANARD DEFL. ON LONGIT. CHARACT. OF BODY WITH CANARD ONLY. BETA=0



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6085) GDLST 603-0 B30C10  
 (AD6083) GDLST 603-0 B30C10  
 (AD6087) GDLST 603-0 B30C10  
 (AD6089) GDLST 603-0 B30C10

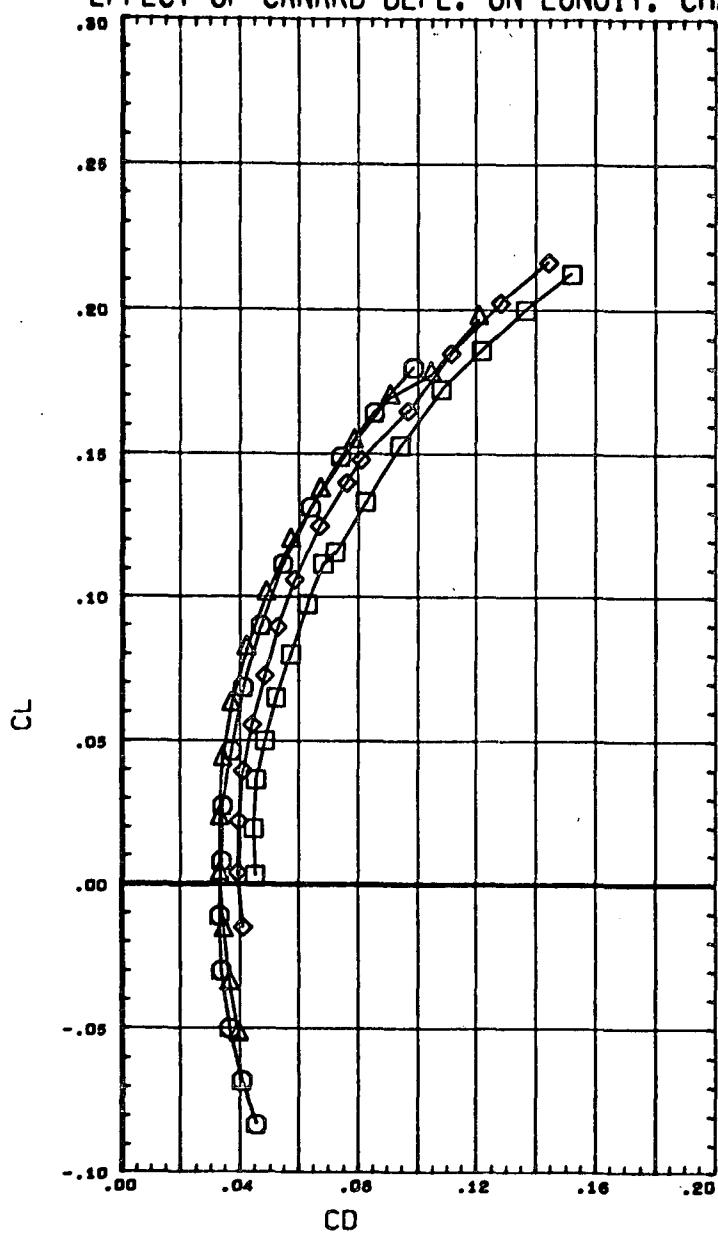
MACH 0.201



BETA CANARD  
 0.000 -10.000  
 0.000 0.000  
 0.000 10.000  
 0.000 15.000

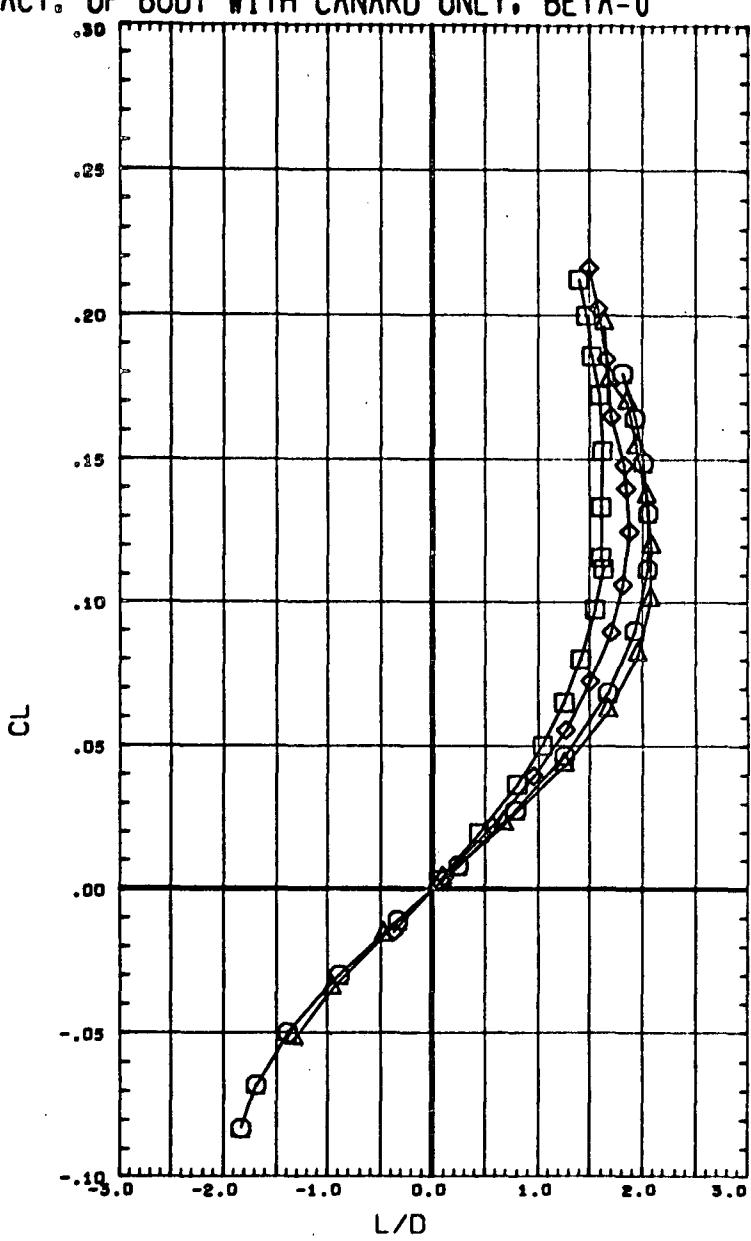
REFERENCE INFORMATION	
SREF	492.4804
LREF	16.1880
BREF	34.6320
XMRP	29.0780
YMRP	0.0000
ZMRP	0.0000
SCALE	0.0200

EFFECT OF CANARD DEFL. ON LONGIT. CHARACT. OF BODY WITH CANARD ONLY, BETA=0



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6085) GDLST 603-0 B30C10  
 (AD6083) GDLST 603-0 B30C10  
 (AD6087) GDLST 603-0 B30C10  
 (AD6089) GDLST 603-0 B30C10

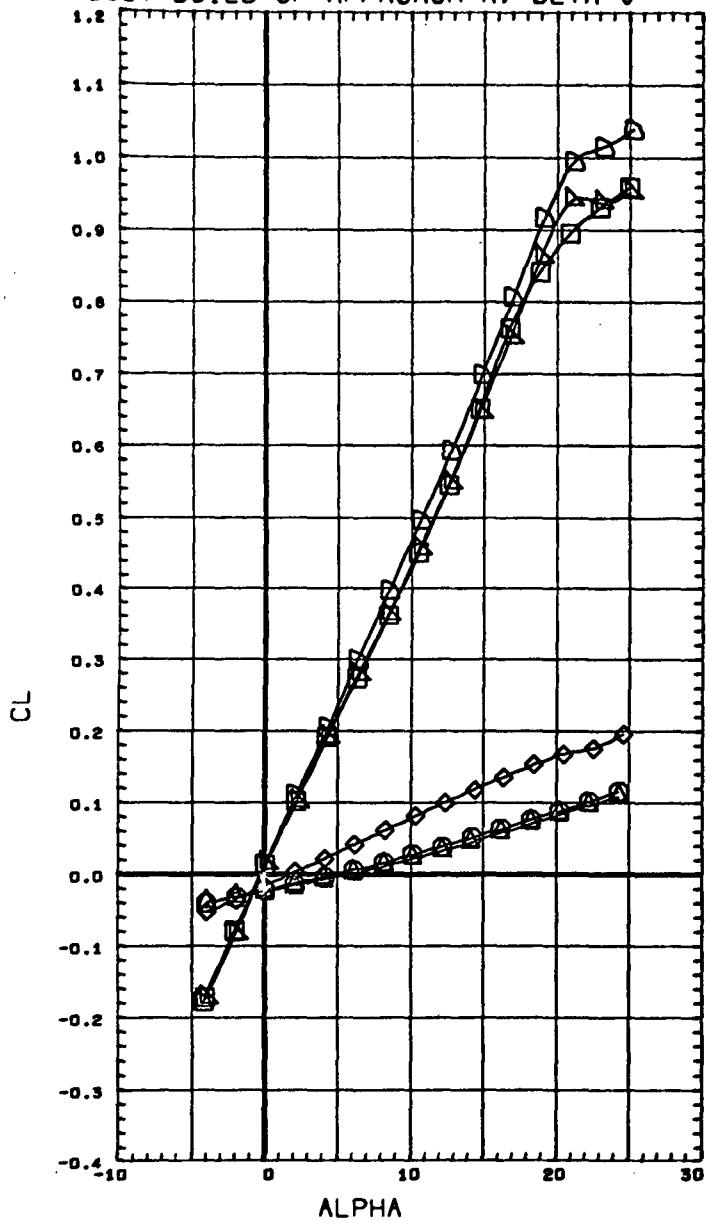
MACH 0.201



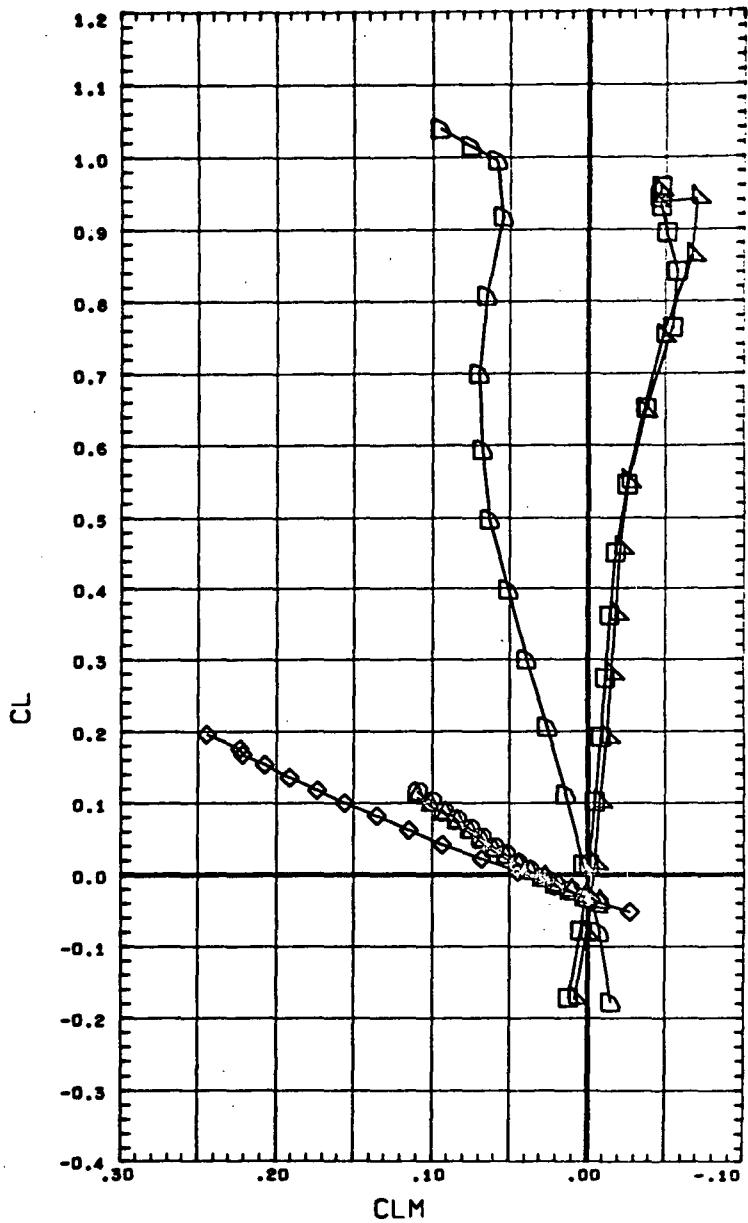
BETA CANARD  
 0.000 -10.000  
 0.000 0.000  
 0.000 10.000  
 0.000 15.000

REFERENCE INFORMATION  
 SREF 492.4804 SQ. IN.  
 LREF 16.1880 IN.  
 BREF 34.6320 IN.  
 XMRP 29.0780 IN.  
 YMRP 0.0000 IN.  
 ZMRP 0.0000 IN.  
 SCALE 0.0200

BODY BUILD-UP APPROACH A, BETA=0

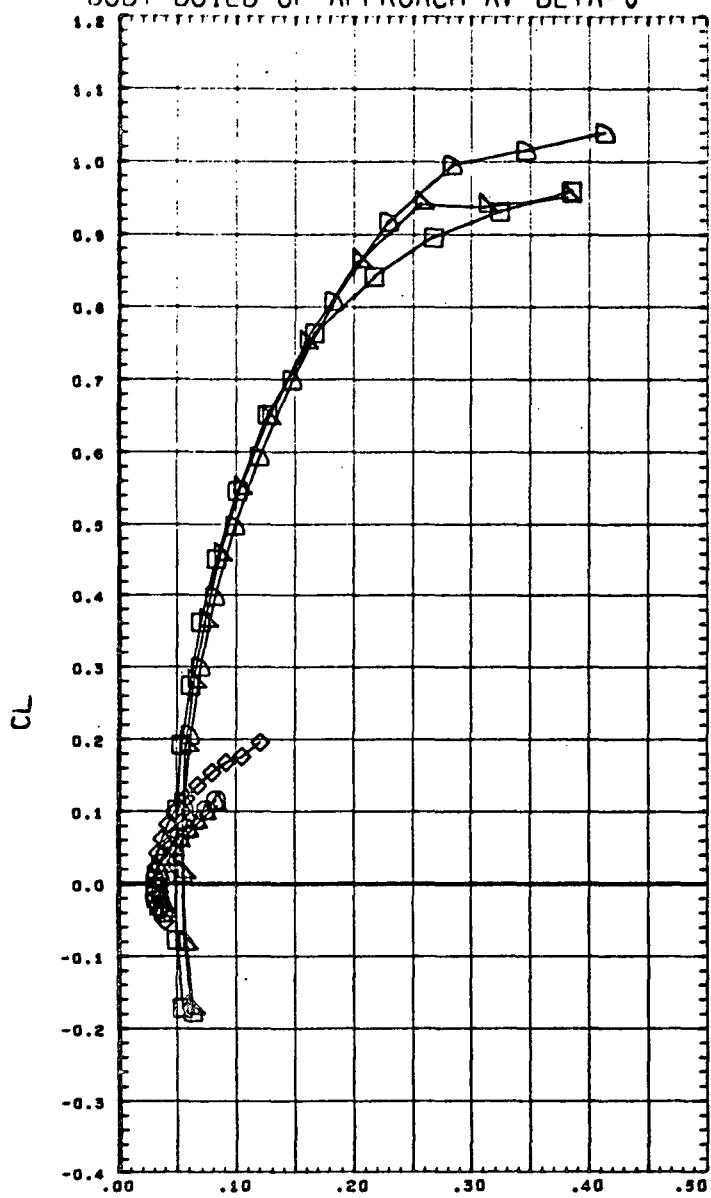


DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6093) Q GDLST 603-0 B30  
 (AD6091) D GDLST 603-0 B30V14  
 (AD6083) D GDLST 603-0 B30C10  
 (AD6077) D GDLST 603-0 B30W23V14  
 (AD6044) D GDLST 603-0 B30W23V14E38  
 (AD6002) D GDLST 603-0 B30W23C10V14E38  
 MACH 0.201



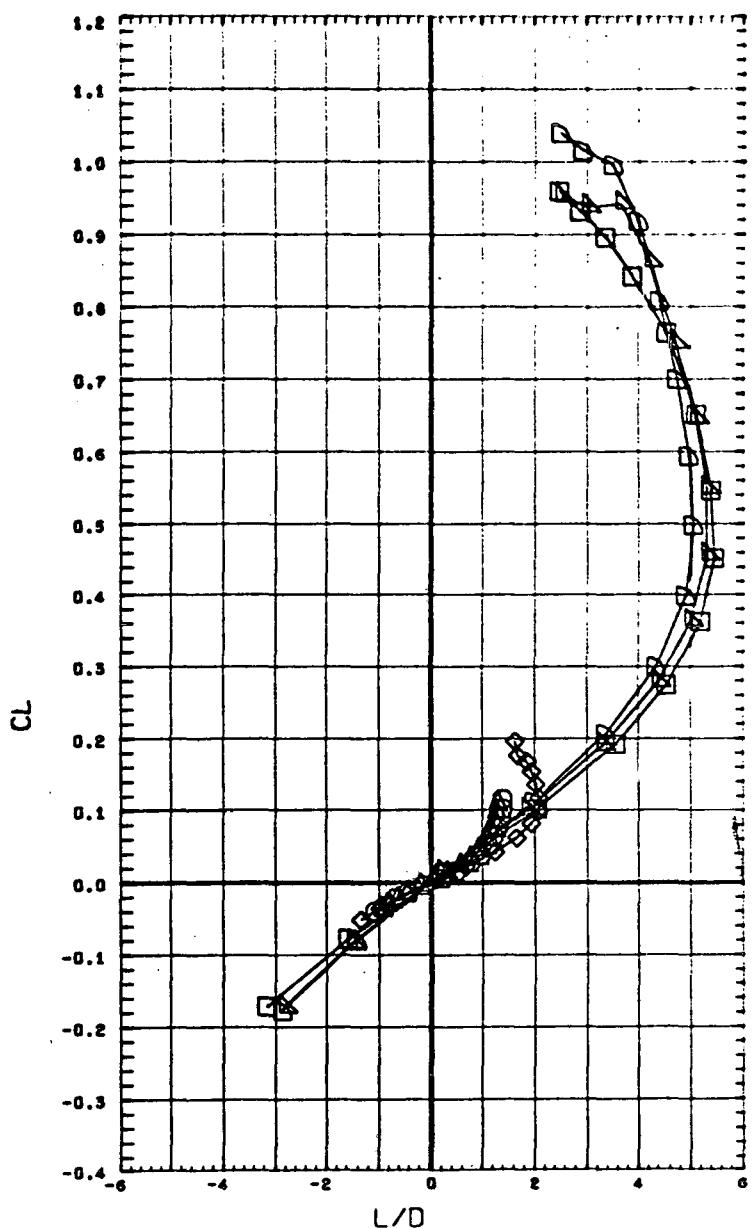
ELEVTR	CANARD	RUDDER	AIRLON	REFERENCE INFORMATION
0.000	0.000	0.000	SREF 492.4804 SQ. IN.	
0.000	0.000	0.000	LREF 16.1880 IN.	
0.000	0.000	0.000	BREF 34.6320 IN.	
0.000	0.000	0.000	XMRP 29.0780 IN.	
0.000	0.000	0.000	YMRP 0.0000 IN.	
0.000	0.000	0.000	ZMRP 0.0000 IN.	
			SCALE 0.0200	

BODY BUILD-UP APPROACH A, BETA=0



DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AC6093)	GCLST 603-0 B30
(AC6091)	GCLST 603-0 B30V14
(AC6083)	GCLST 603-0 B30C10
(AC6077)	GCLST 603-0 B30W23V14
(AC6044)	GCLST 603-0 B30W23V14E38
(AC6002)	GCLST 603-0 B30W23C10V14E38

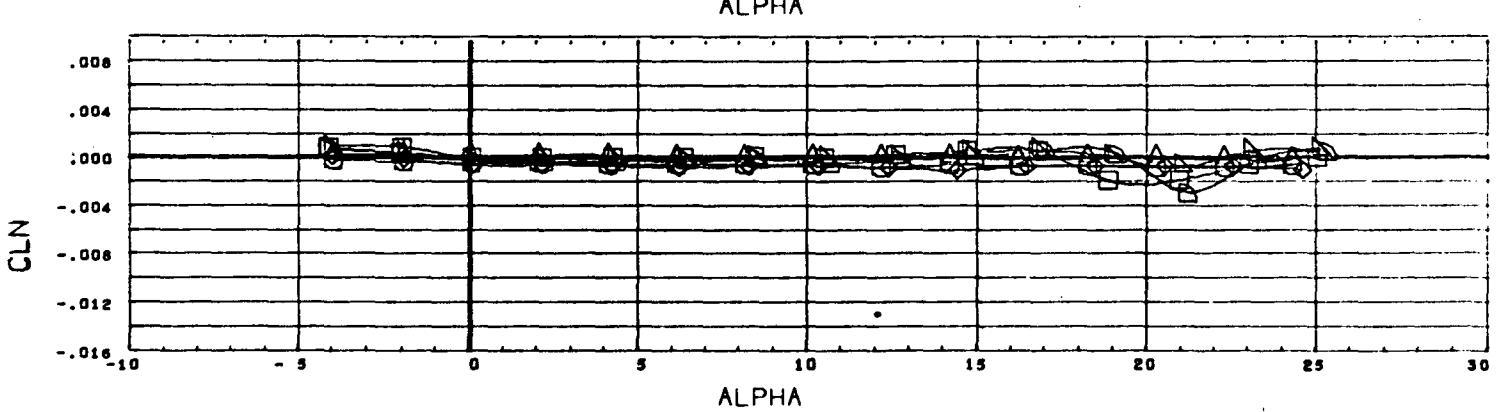
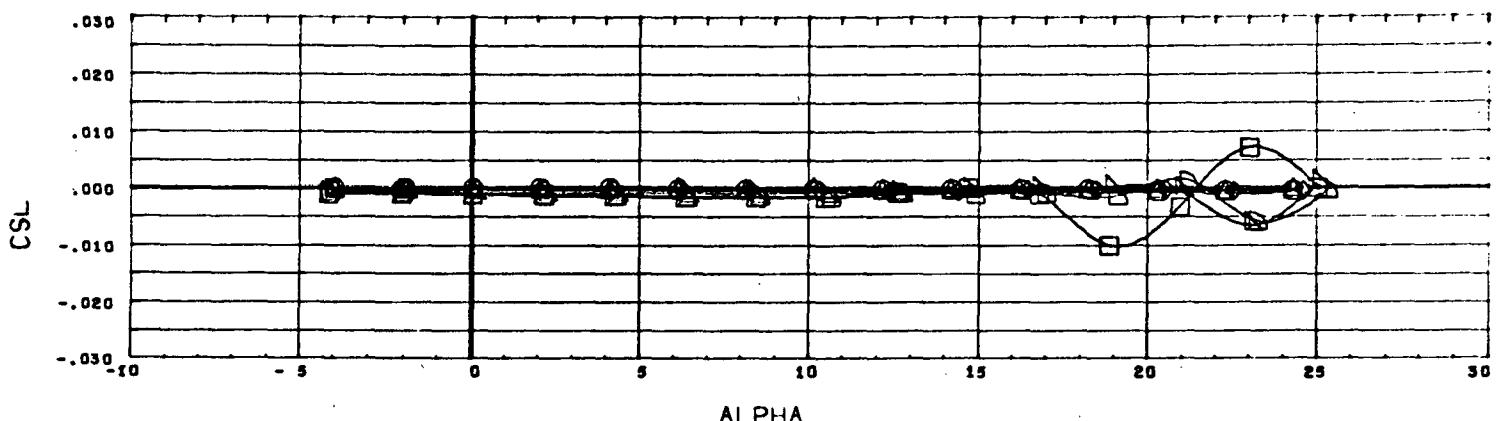
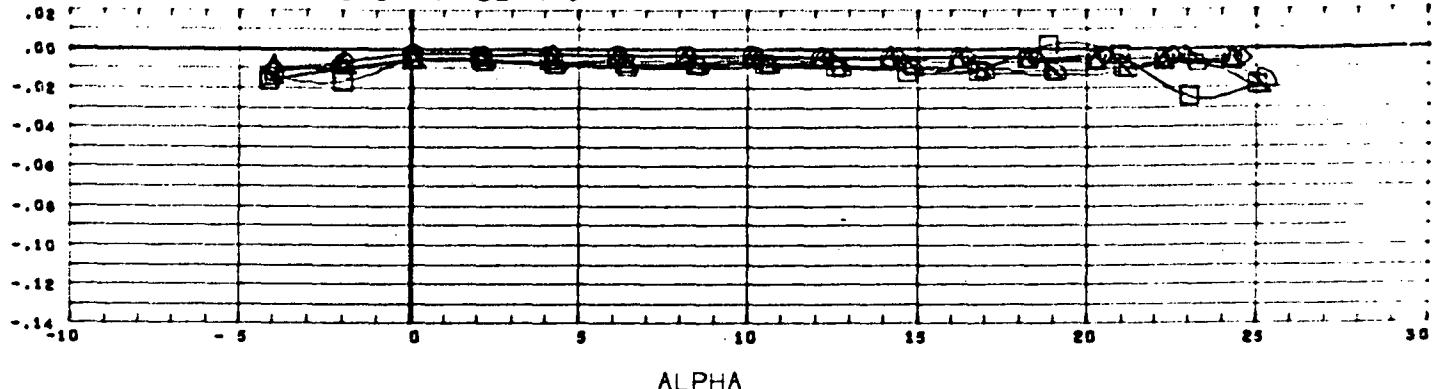
MACH .20



ELEVTR	CANARD	RUDER	AIRLON	REFERENCE INFORMATION
		0.000		SREF 492.4804 SQ. IN.
				LREF 16.1880 IN.
				BREF 34.6320 IN.
0.000		0.000	0.000	XMRP 29.9780 IN.
0.000		0.000	0.000	YMRP 0.0000 IN.
0.000	0.000	0.000	0.000	ZMRP 0.0000 IN.
				SCALE 0.0200

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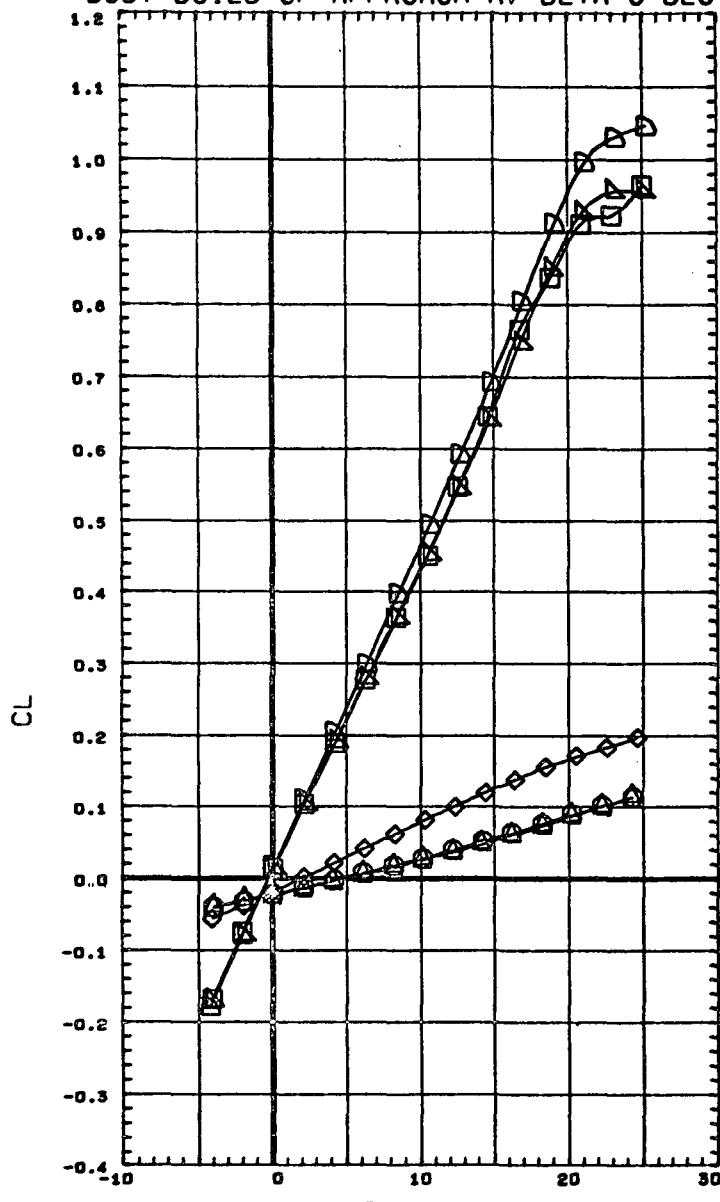
BODY BUILD-UP APPROACH A, BETA=0



DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AD6093)	GCLST 603-0 B30
(AD6091)	GCLST 603-0 B30V14
(AD6083)	GCLST 603-0 B30C10
(AD6077)	GCLST 603-0 B30W23V14
(AD6044)	GCLST 603-0 B30W23V14E38
(AD6002)	GCLST 603-0 B30W23C10V14E38

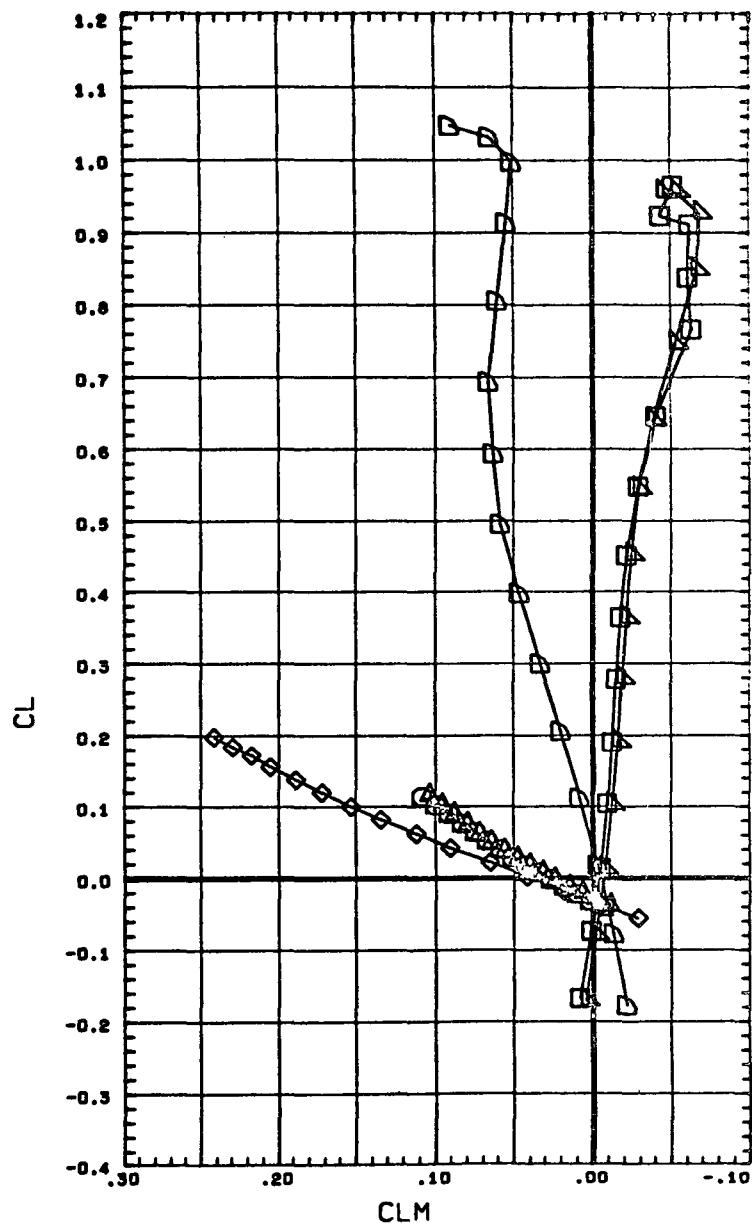
	ELEVTR	CANARD	RUDDER	AILRON	REFERENCE INFORMATION
			0.000		SREF 492.4804 SQ. IN.
		0.000			LREF 16.1880 IN.
			0.000		BREF 34.6320 IN.
				XMRP 29.0780 IN.	
				YMRP 0.0000 IN.	
				ZMRP 0.0000 IN.	
				SCALE 0.0200	

BODY BUILD-UP APPROACH A, BETA=5 DEG



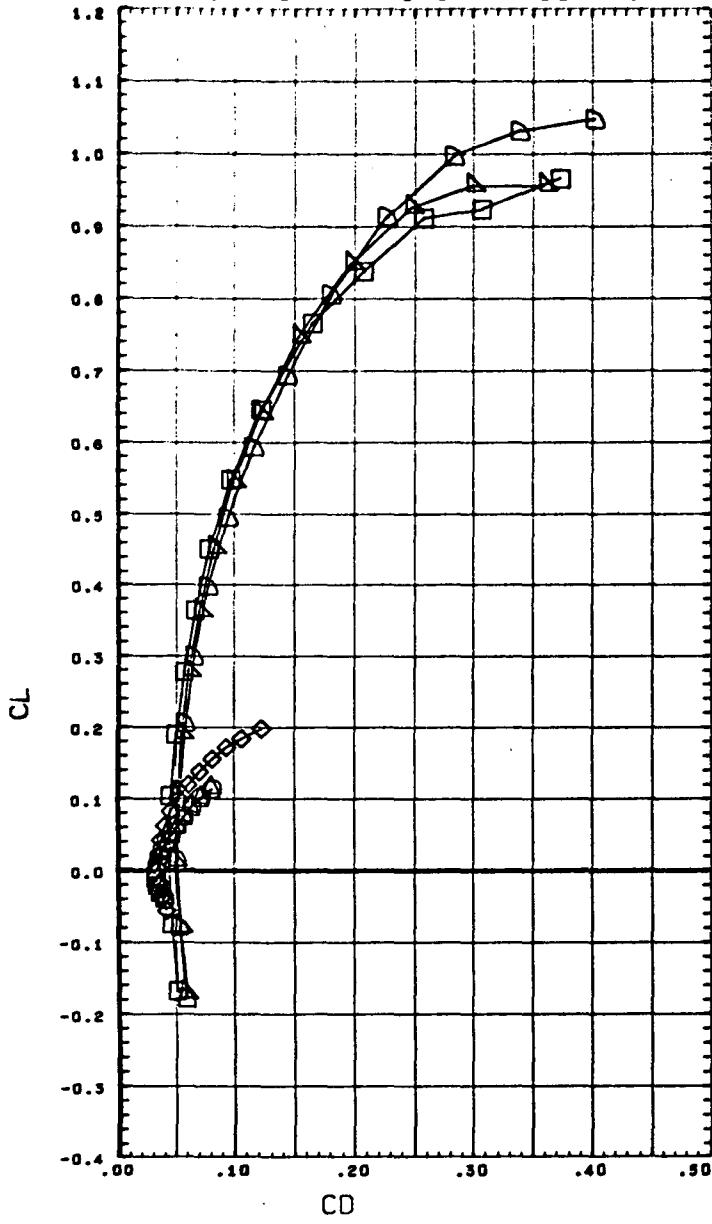
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AD6094)	GDLST 603-0 B30
(AD6092)	GDLST 603-0 B30V14
(AD6084)	GDLST 603-0 B30C10
(AD6078)	GDLST 603-0 B30W23V14
(AD6045)	GDLST 603-0 B30W23V14E38
(AD6003)	GDLST 603-0 B30W23C10V14E38

MACH 0.201

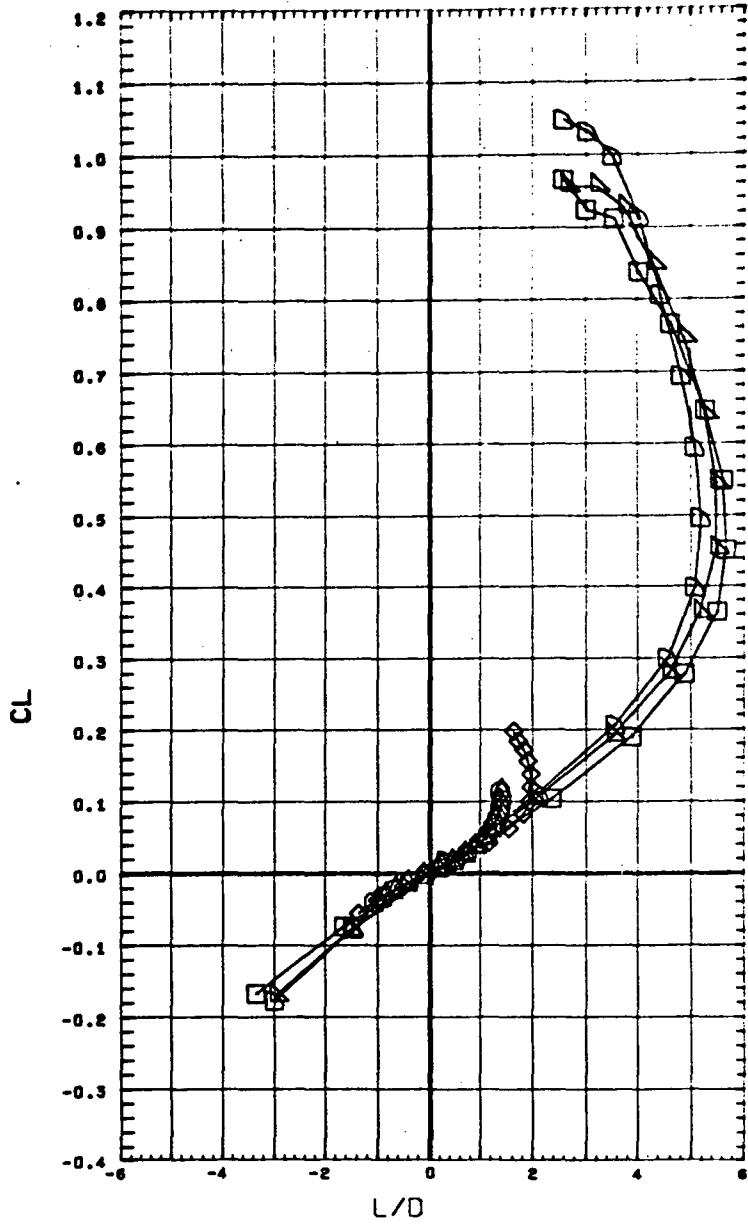


ELEVTR	CANARD	RUDDER	AIRLON	REFERENCE INFORMATION
	0.000	0.000	0.000	SREF 492.4804 SQ. IN.
0.000	0.000	0.000	0.000	LREF 16.1880 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
0.000	0.000	0.000	0.000	XMRP 29.0780 IN.
0.000	0.000	0.000	0.000	YMRP 0.0000 IN.
				ZMRP 0.0000 IN.
				SCALE 0.0200

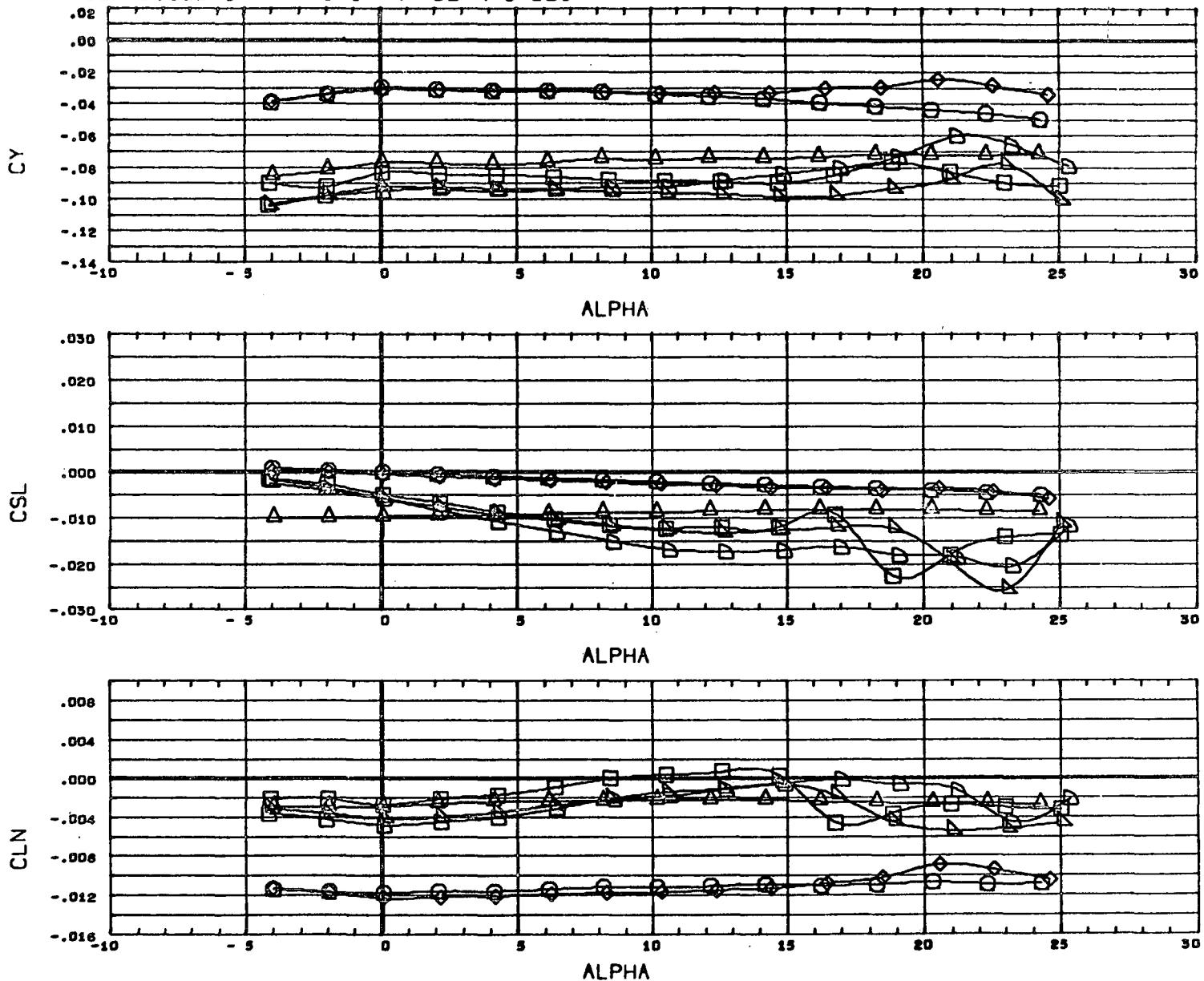
BODY BUILD-UP APPROACH A, BETA=5 DEG



MACH .20



BODY BUILD-UP APPROACH A, BETA=5 DEG

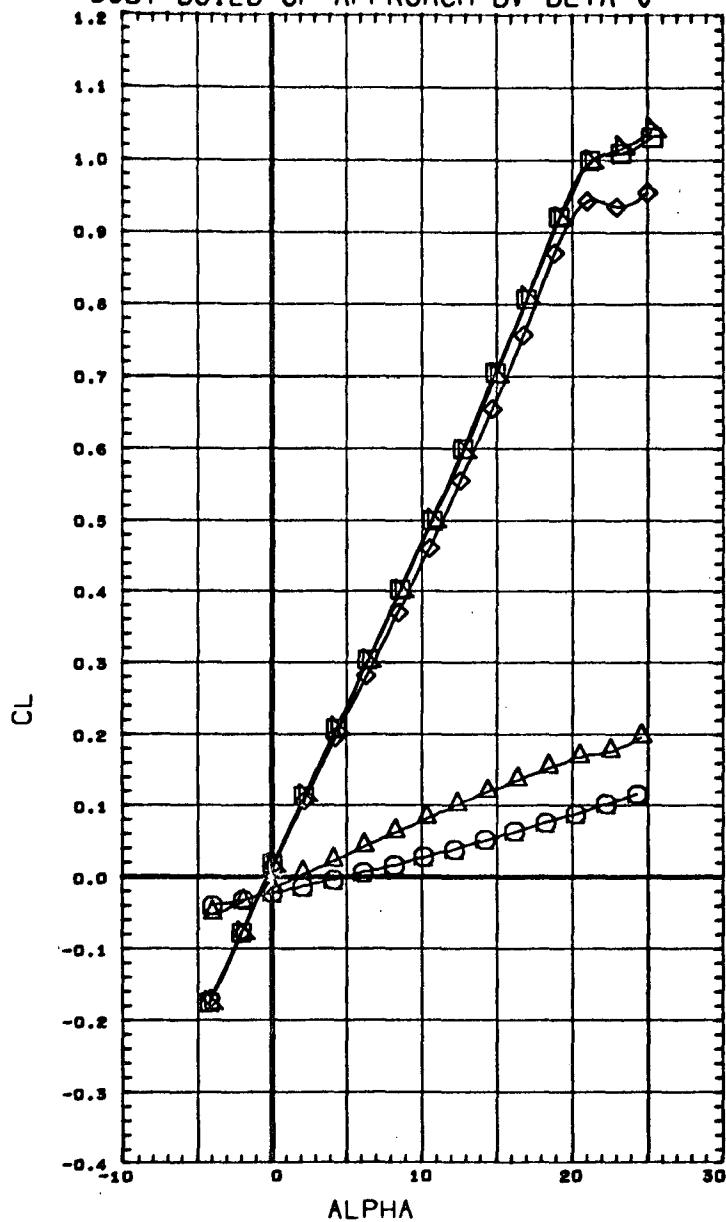


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AD6094)	GDLST 603-0 B30
(AD6092)	GDLST 603-0 B30V14
(AD6084)	GDLST 603-0 B30C10
(AD6078)	GDLST 603-0 B30W23V14
(AD6045)	GDLST 603-0 B30W23V14E38
(AD6003)	GDLST 603-0 B30W23C10V14E38

MACH 0.201

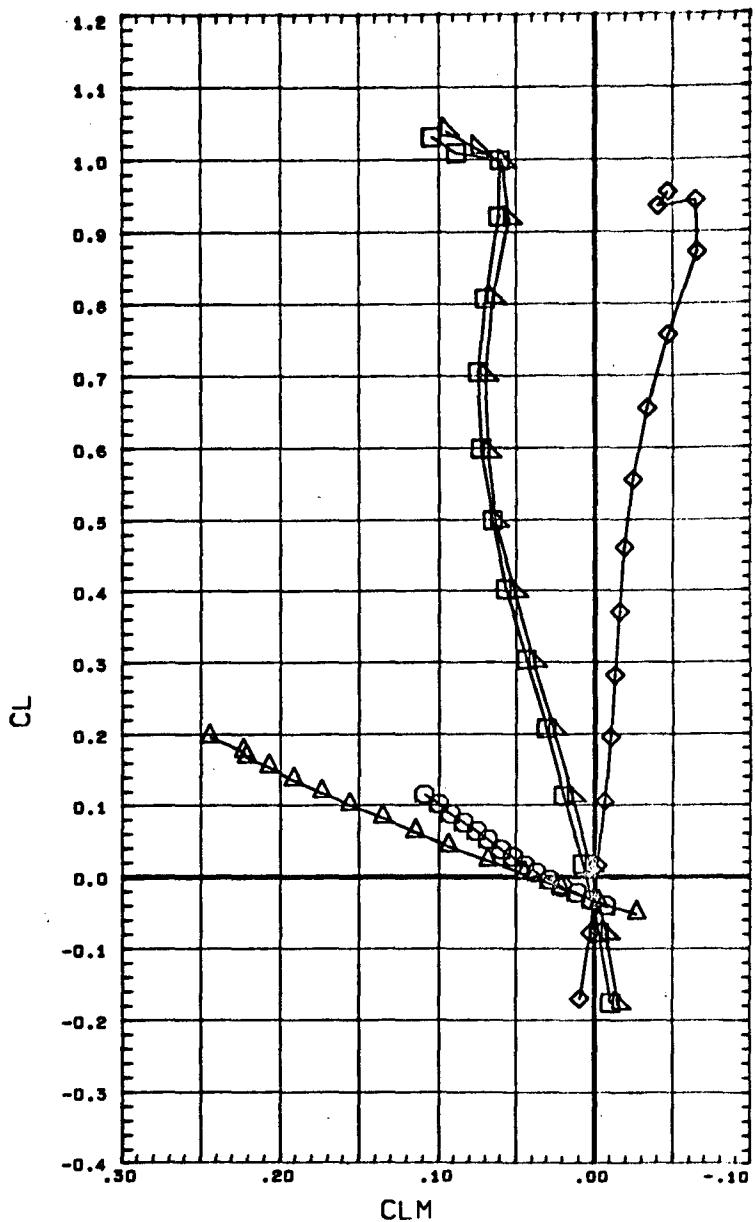
ELEVTR	CANARD	RUDDER	AIRLON	REFERENCE INFORMATION
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0.000	0.000	0.000	0.000	LREF 16.1880 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
0.000	0.000	0.000	0.000	XMRP 29.0780 IN.
0.000	0.000	0.000	0.000	YMRP 0.0000 IN.
0.000	0.000	0.000	0.000	ZMRP 0.0000 IN.
				SCALE 0.0200

BODY BUILD-UP APPROACH B, BETA=0



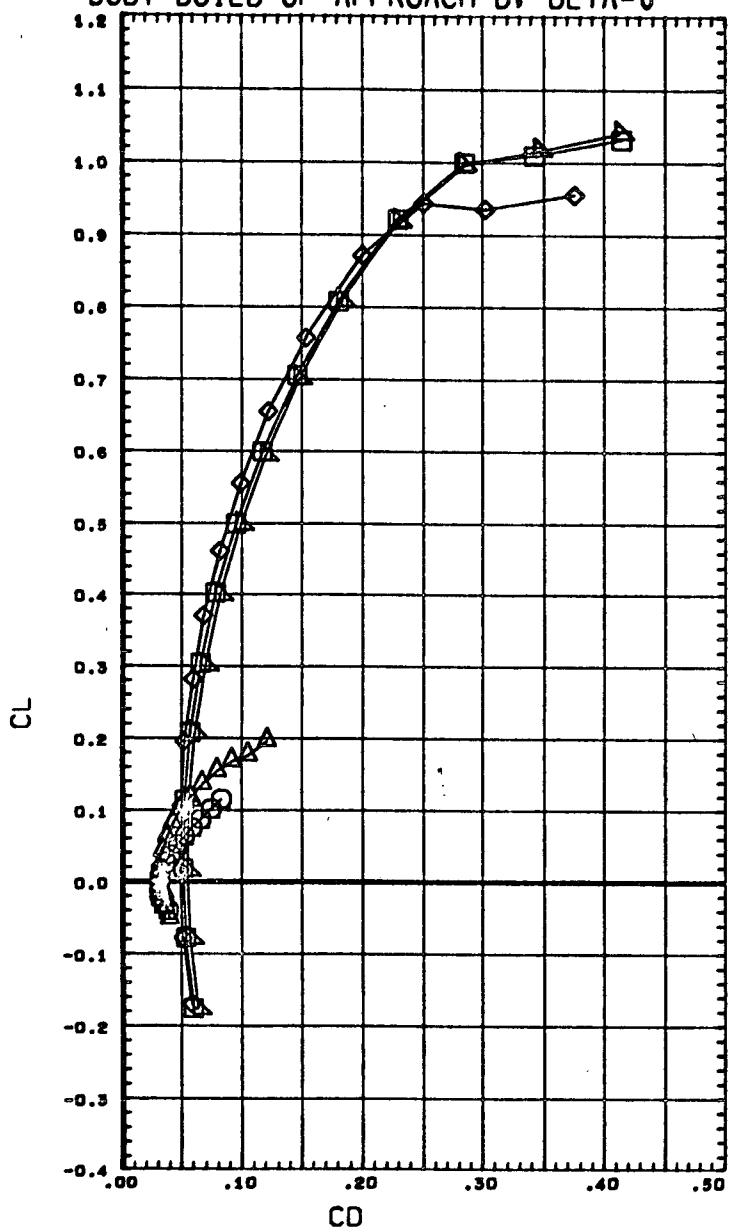
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6093) GDLST 603-0 B30  
 (AD6083) GDLST 603-0 B30C10  
 (AD6085) GDLST 603-0 B30W23E38  
 (AD6054) GDLST 603-0 B30W23C10E38  
 (AD6002) GDLST 603-0 B30W23C10V14E38

MACH 0.201



ELEVTR	CANARD	RUDDER	AIRLON	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 492.4804 SQ. IN.
0.000	0.000	0.000	0.000	LREF 16.1880 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
0.000	0.000	0.000	0.000	XMRP 29.0780 IN.
0.000	0.000	0.000	0.000	YMRP 0.0000 IN.
0.000	0.000	0.000	0.000	ZMRP 0.0000 IN.
				SCALE 0.0200

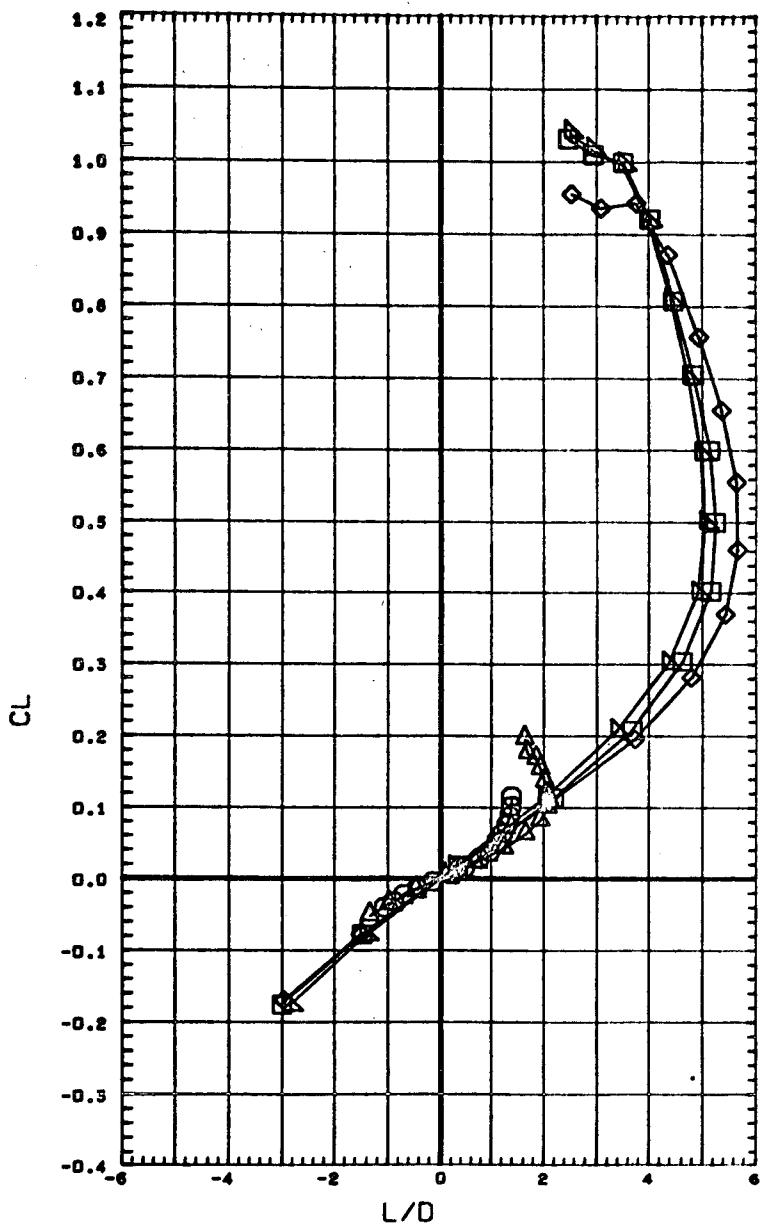
BODY BUILD-UP APPROACH B, BETA=0



DATA SET SYMBOL CONFIGURATION DESCRIPTION

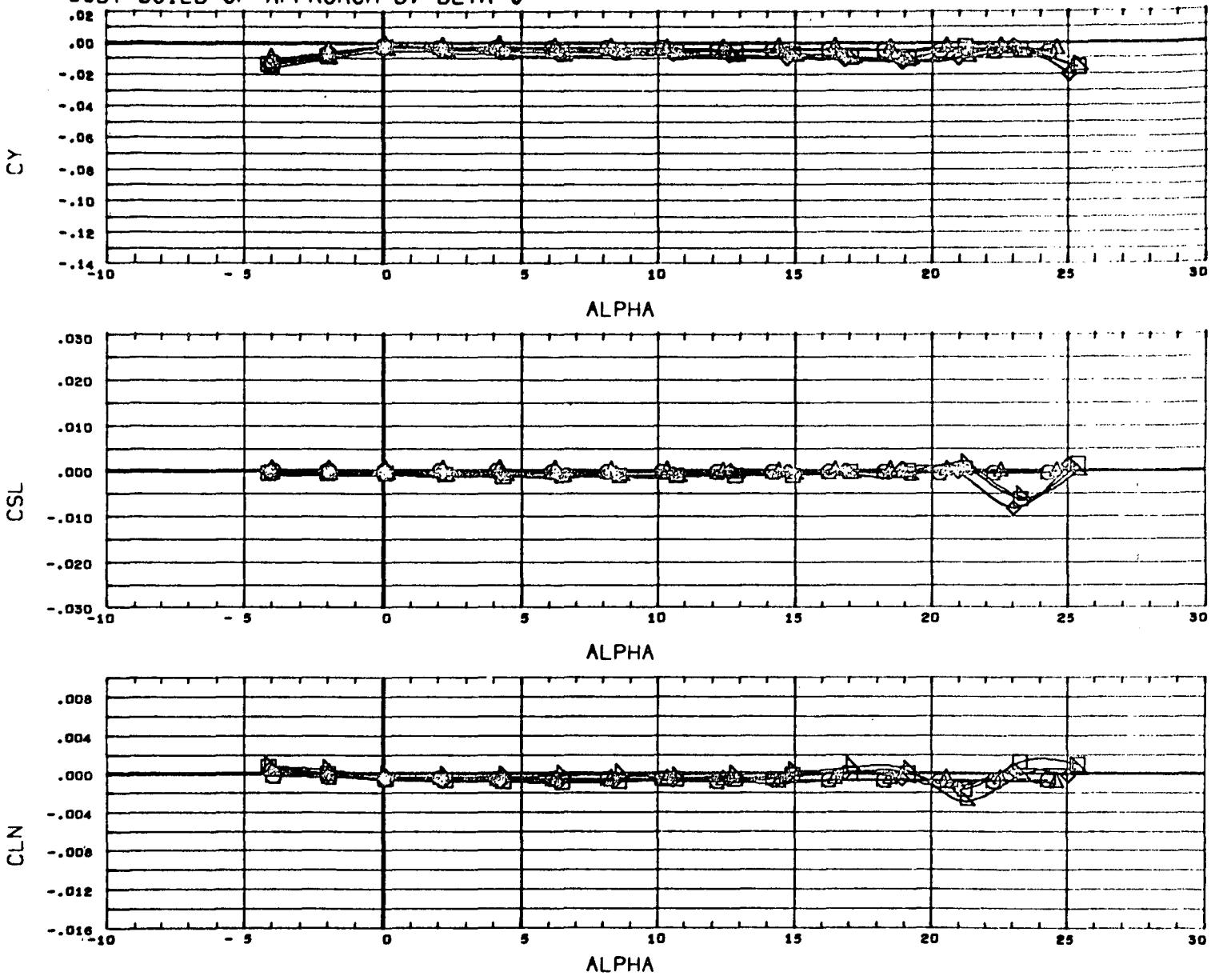
(AD6093)	○	GDLST 603-0	B30
(AD6083)	△	GDLST 603-0	B30C10
(AD6065)	◇	GDLST 603-0	B30W23E38
(AD6054)	□	GDLST 603-0	B30W23C10E38
(AD6002)	▽	GDLST 603-0	B30W23C10V14E38

MACH 0.201



ELEVTR	CANARD	RUDDER	AIRLON	REFERENCE INFORMATION
0.000	0.000			SREF 492.4804 SQ. IN.
0.000	0.000	0.000		LREF 16.1880 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
0.000	0.000	0.000	0.000	XMRP 29.0780 IN.
				YMRP 0.0000 IN.
				ZMRP 0.0000 IN.
				SCALE 0.0200

BODY BUILD-UP APPROACH B, BETA=0

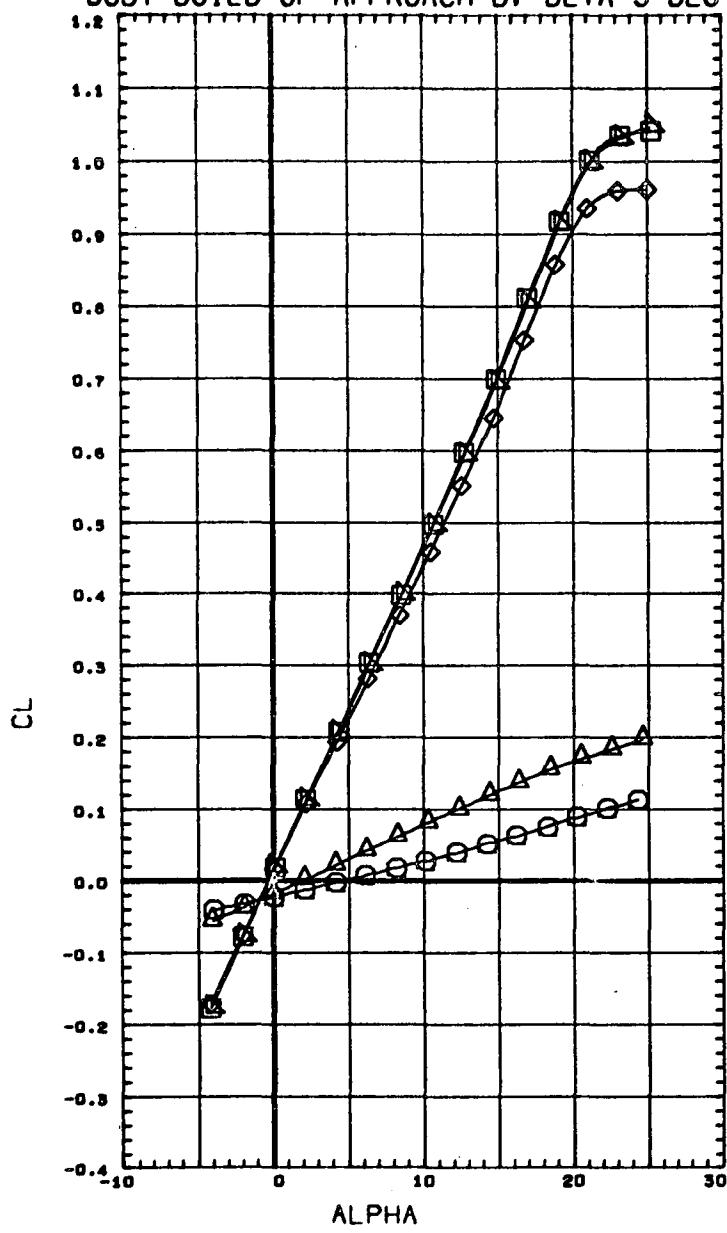


DATA SET	SYMBOL	CONFIGURATION	DESCRIPTION
(AD6093)		GDLST 603-0	B30
(AD6083)		GDLST 603-0	B30C10
(AD6065)		GDLST 603-0	B30W23E38
(AD6034)		GDLST 603-0	B30W23C10E38
(AD6002)		GDLST 603-0	B30W23C10V14E38

ELEVTR	CANARD	RUDDER	AIRLON	REFERENCE	INFORMATION
	0.000			SREF	492.4804 SQ. IN.
		0.000		LREF	16.1680 IN.
			0.000	BREF	34.6320 IN.
			0.000	XMRP	29.0780 IN.
			0.000	YMRP	0.0000 IN.
			0.000	ZMRP	0.0000 IN.
				SCALE	0.0200

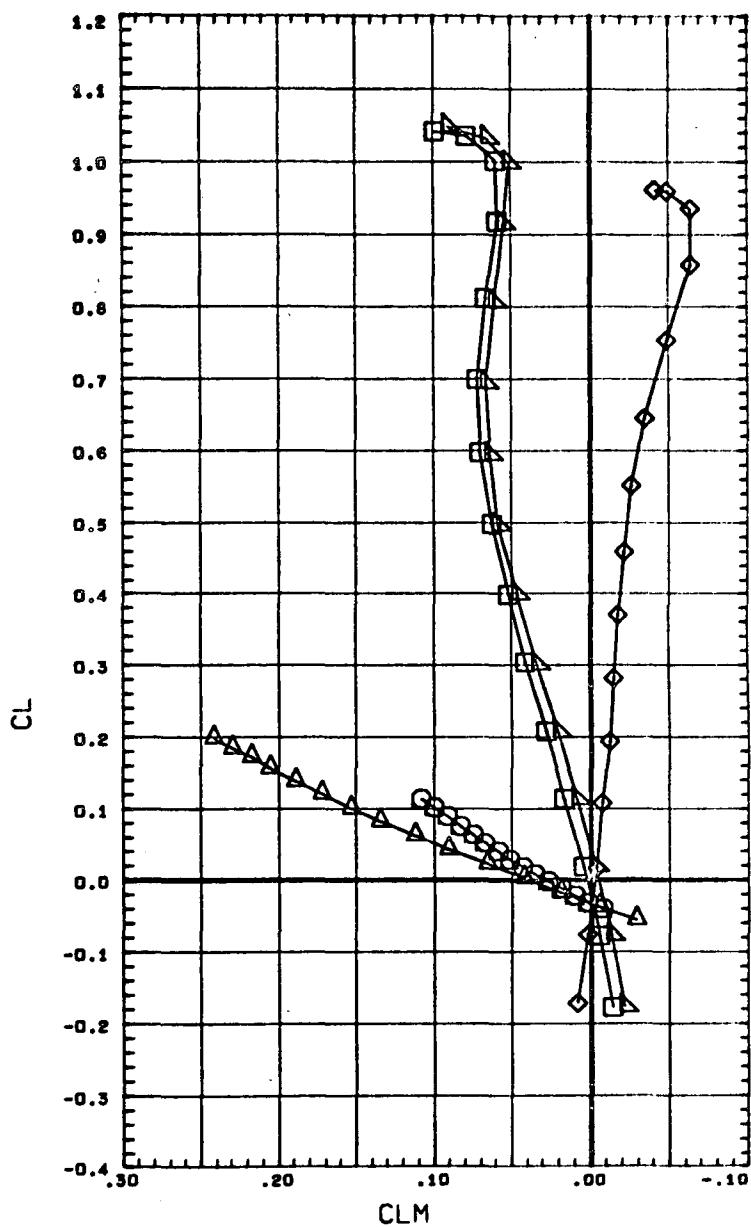
MACH 0.201

BODY BUILD-UP APPROACH B, BETA=5 DEG



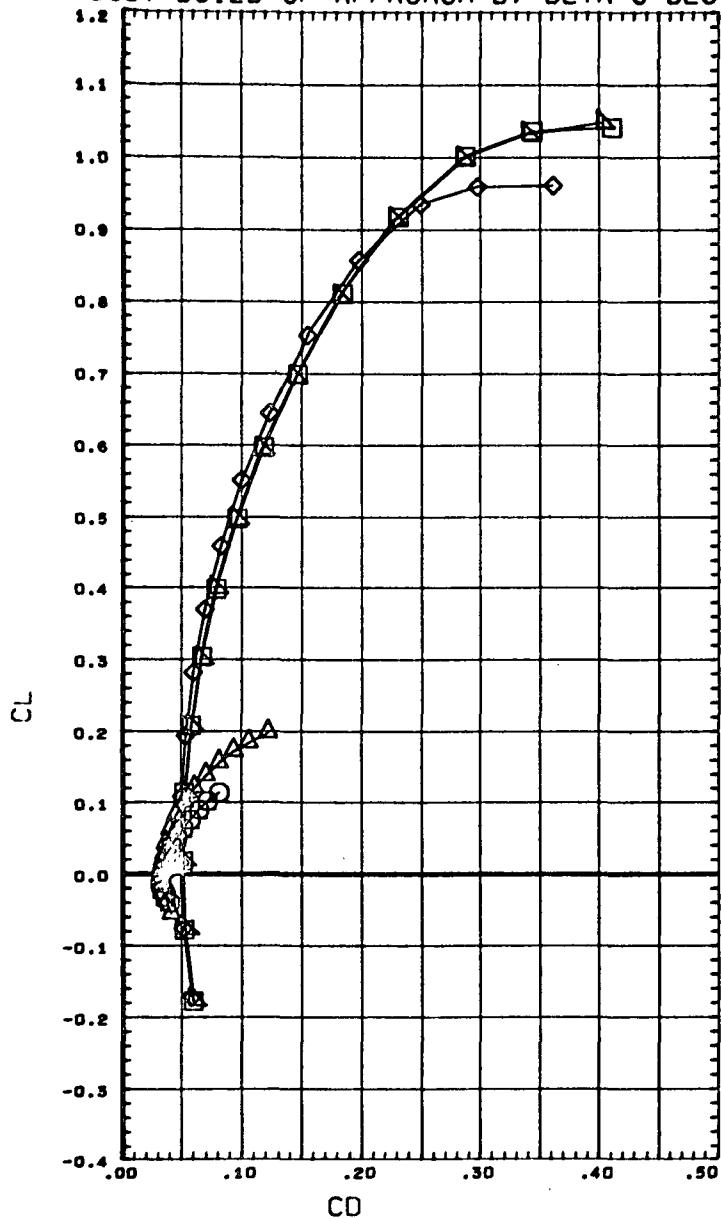
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 (AD6094) GDLST 603-0 B30  
 (AD6084) GDLST 603-0 B30C10  
 (AD6066) GDLST 603-0 B30W23E38  
 (AD6055) GDLST 603-0 B30W23C10E38  
 (AD6003) GDLST 603-0 B30W23C10V14E38

MACH 0.201



ELEVTR	CANARD	RUDER	AIRLON	REFERENCE INFORMATION
0.000	0.000			SREF 492.4804 SQ. IN.
0.000	0.000	0.000	BREF 16.1880 IN.	
0.000	0.000	0.000	XMRP 29.0780 IN.	
0.000	0.000	0.000	YMRP 0.0000 IN.	
			ZMRP 0.0000 IN.	
			SCALE 0.0200	

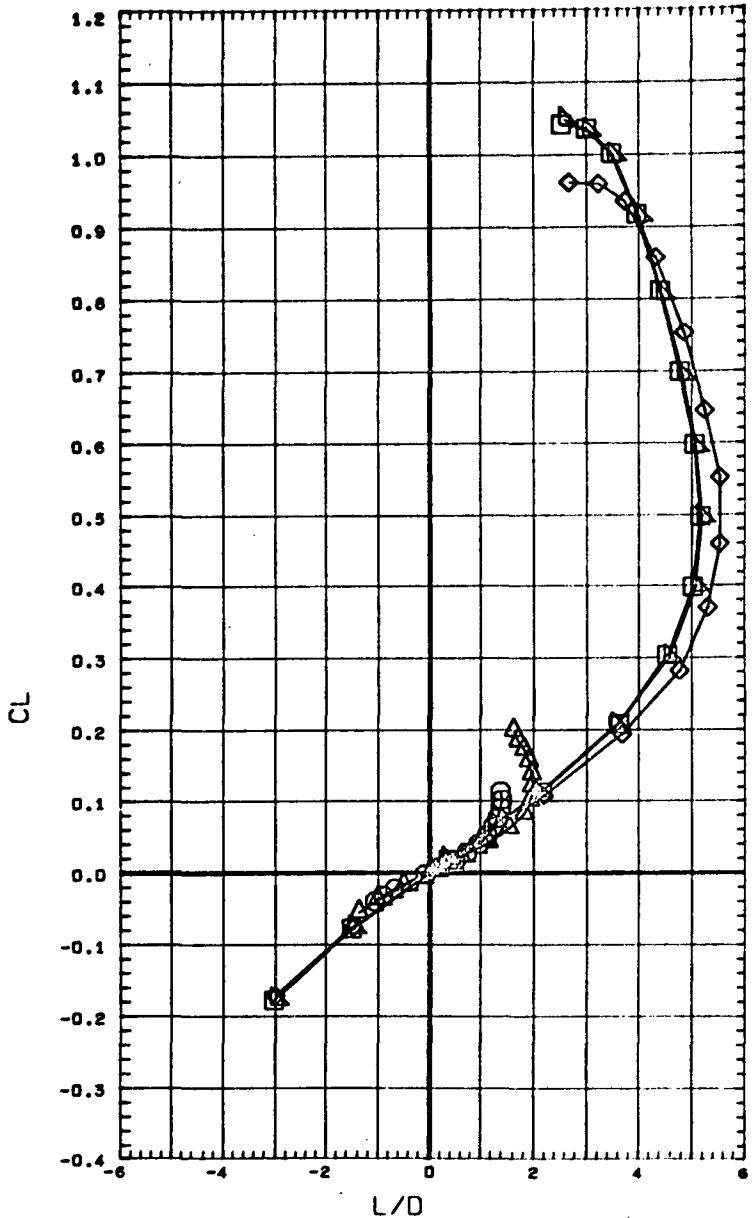
BODY BUILD-UP APPROACH B, BETA=5 DEG



DATA SET SYMBOL CONFIGURATION DESCRIPTION

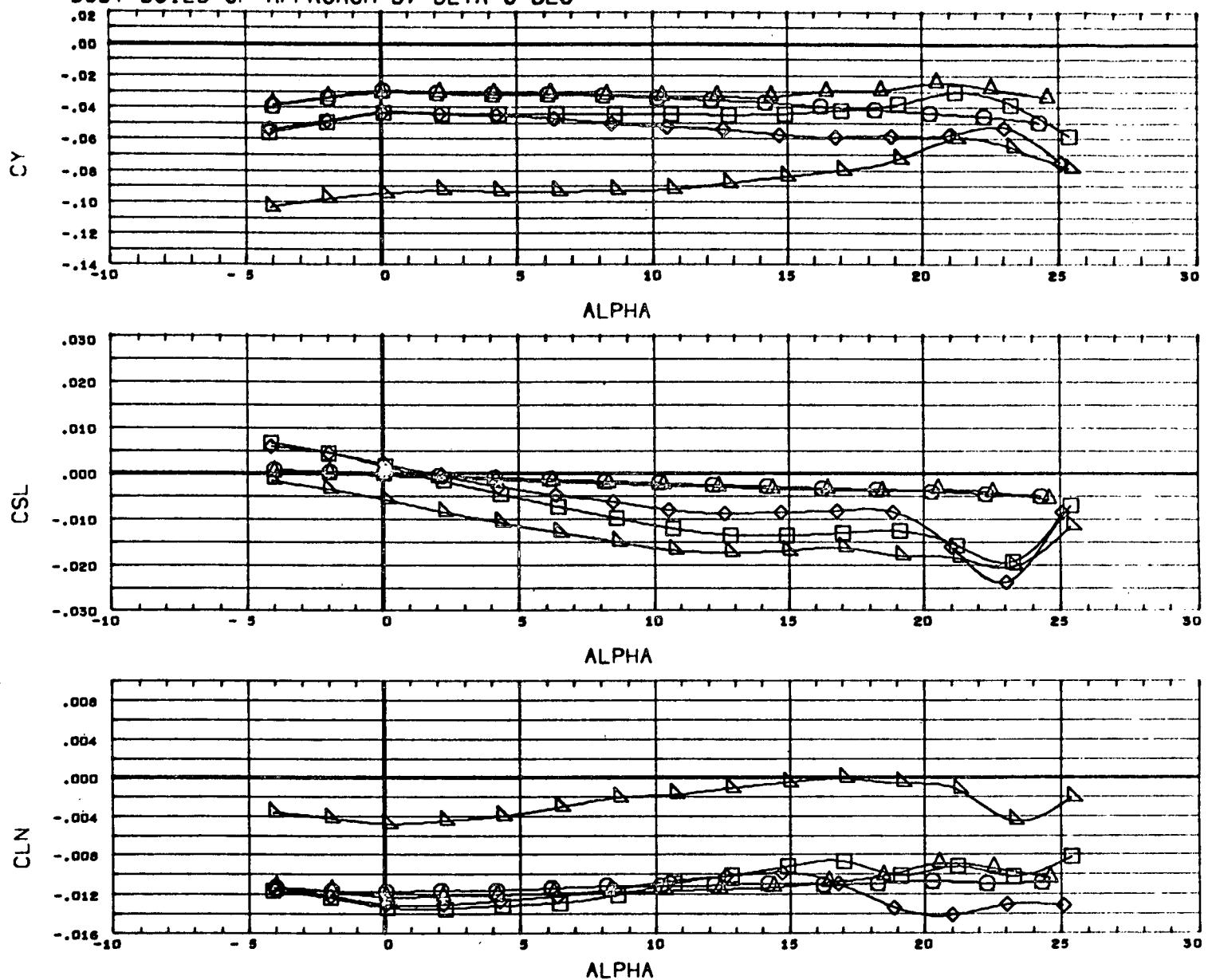
(AD6094)	(AD6084)	(AD6066)	(AD6055)	(AD6003)
○	□	◇	△	▽
GDLST 603-0	GDLST 603-0	GDLST 603-0	GDLST 603-0	GDLST 603-0
B30	B30C10	B30W23E38	B30W23C10E38	B30W23C10V14E38

MACH 0.201



ELEVTR	CANARD	RUDDER	AILRON	REFERENCE INFORMATION
	0.000		0.000	SREF 492.4804 SQ. IN.
0.000	0.000	0.000	0.000	LREF 16.1880 IN.
0.000	0.000	0.000	0.000	BREF 34.6320 IN.
0.000	0.000	0.000	0.000	XMRP 29.0780 IN.
0.000	0.000	0.000	0.000	YMRP 0.0000 IN.
0.000	0.000	0.000	0.000	ZMRP 0.0000 IN.
				SCALE 0.0200

BODY BUILD-UP APPROACH B, BETA=5 DEG



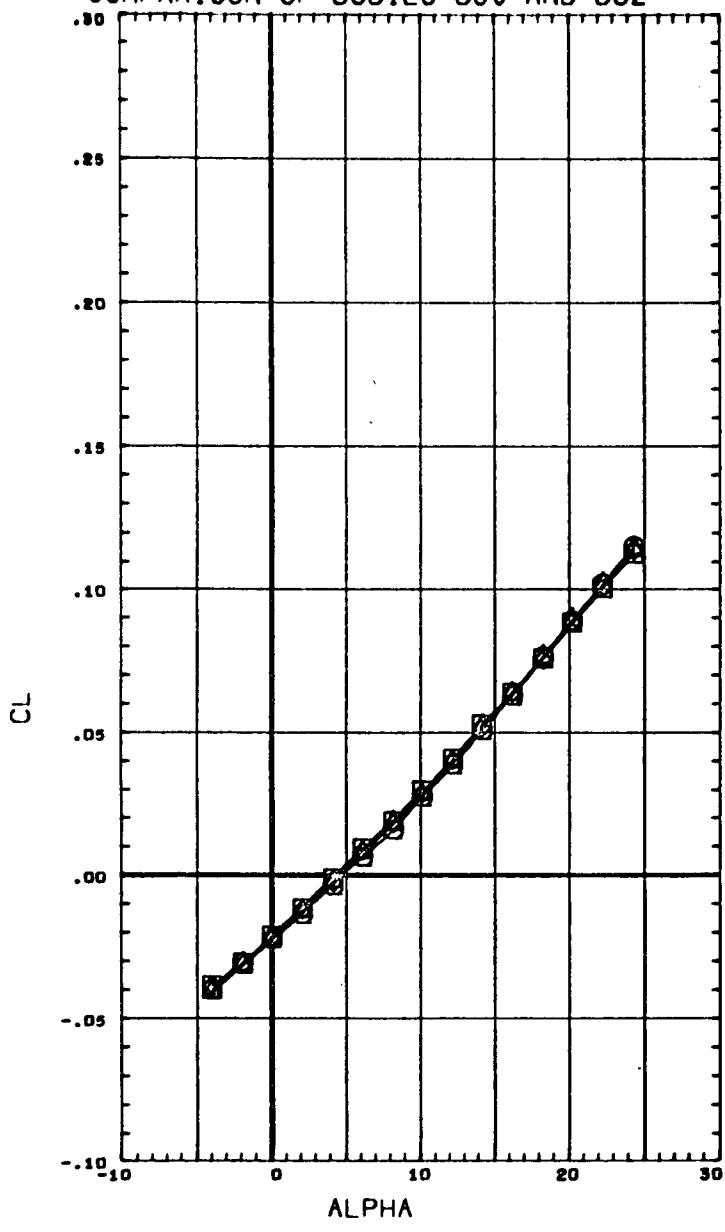
DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AD6094)		GDLST 603-0	B30
(AD6084)		GDLST 603-0	B30C10
(AD6066)		GDLST 603-0	B30W23E38
(AD6055)		GDLST 603-0	B30W23C10E38
(AD6003)		GDLST 603-0	B30W23C10V14E38

	ELEVTR	CANARD	RUDDER	AILRON	REFERENCE INFORMATION
SREF	492.4804	sq. IN.			
LREF	16.1880	IN.			
BREF	34.6320	IN.			
XHMP	29.0780	IN.			
YHMP	0.0000	IN.			
ZHMP	0.0000	IN.			
SCALE	0.0200				

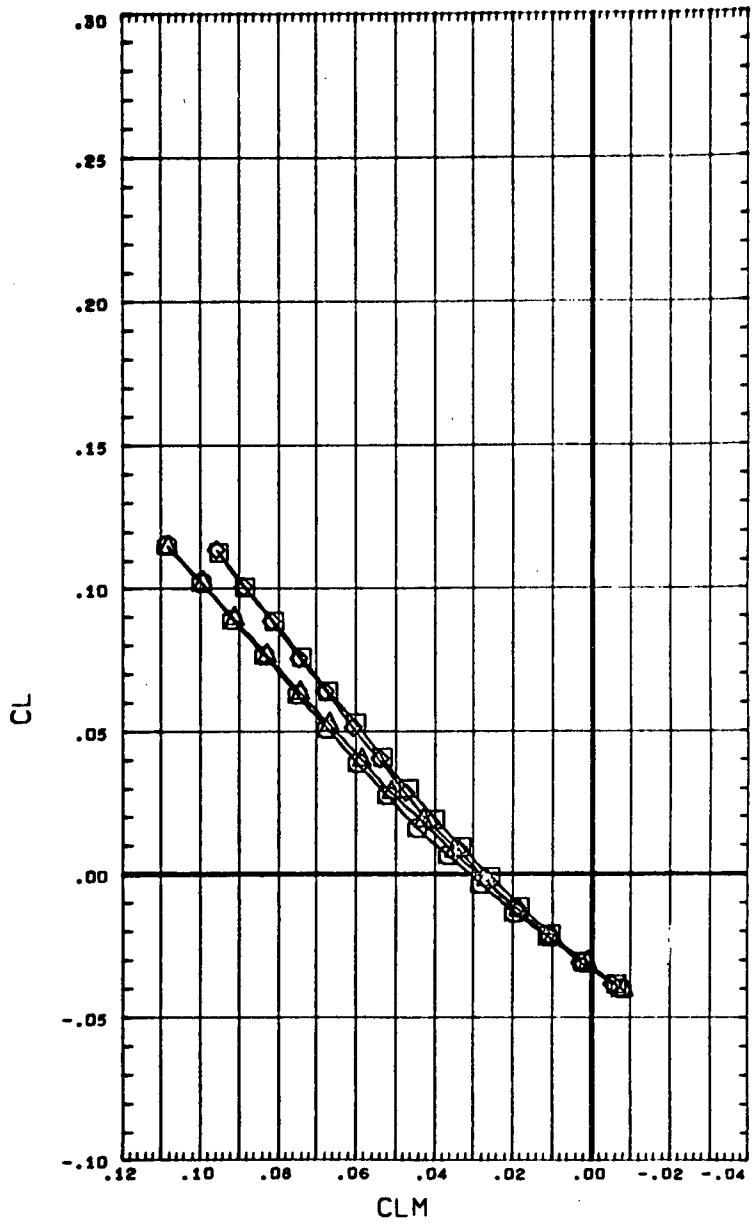
MACH 0.201

## COMPARISON OF BODIES B30 AND B32



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6093) Q GDLST 603-0 B30  
 (AD6094) X GDLST 603-0 B30  
 (AD6095) D GDLST 603-0 B32  
 (AD6096) C GDLST 603-0 B32

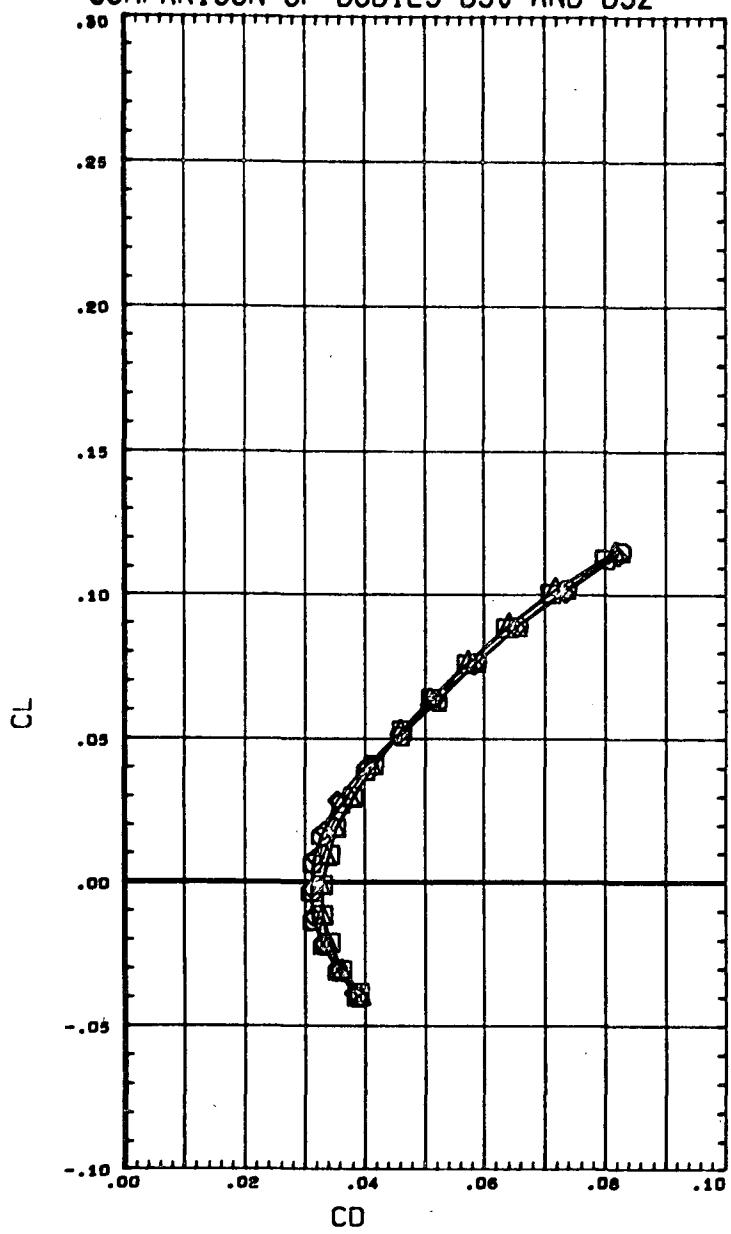
MACH 0.201



BETA CANARD  
 0.000  
 5.000  
 0.000  
 5.000

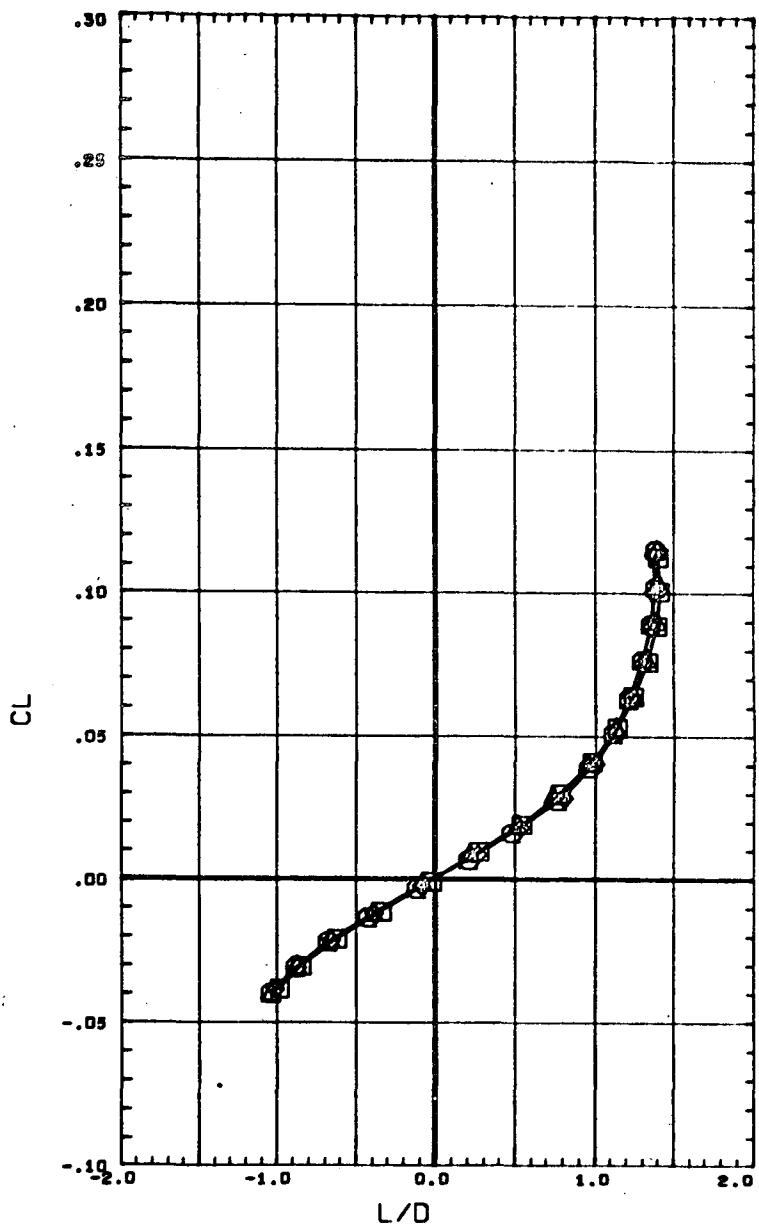
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 LREF 16.1880 IN.  
 BREF 34.6320 IN.  
 XHRP 29.0780 IN.  
 YMRP 0.0000 IN.  
 ZMRP 0.0000 IN.  
 SCALE 0.0200

# COMPARISON OF BODIES B30 AND B32



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6093) GDLST 603-0 B30  
 (AD6094) GDLST 603-0 B30  
 (AD6095) GDLST 603-0 B32  
 (AD6096) GDLST 603-0 B32

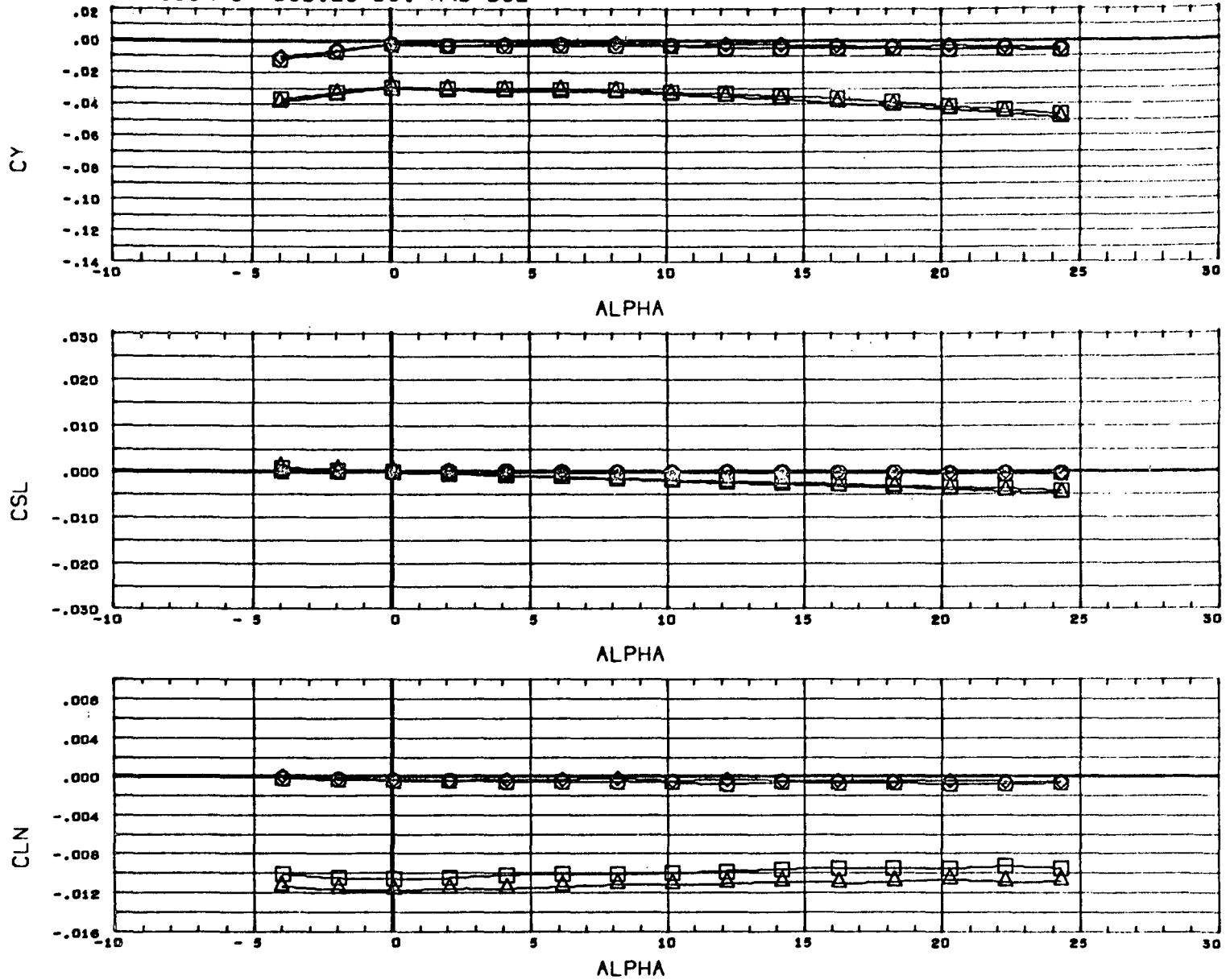
MACH 0.201



BETA CANARD  
 0.000  
 5.000  
 0.000  
 5.000 .

REFERENCE INFORMATION  
 SREF 492.4804 SQ. IN.  
 LREF 16.1880 IN.  
 BREF 34.6320 IN.  
 XMRP 29.0780 IN.  
 YMRP 0.0000 IN.  
 ZMRP 0.0000 IN.  
 SCALE 0.0200

# COMPARISON OF BODIES B30 AND B32



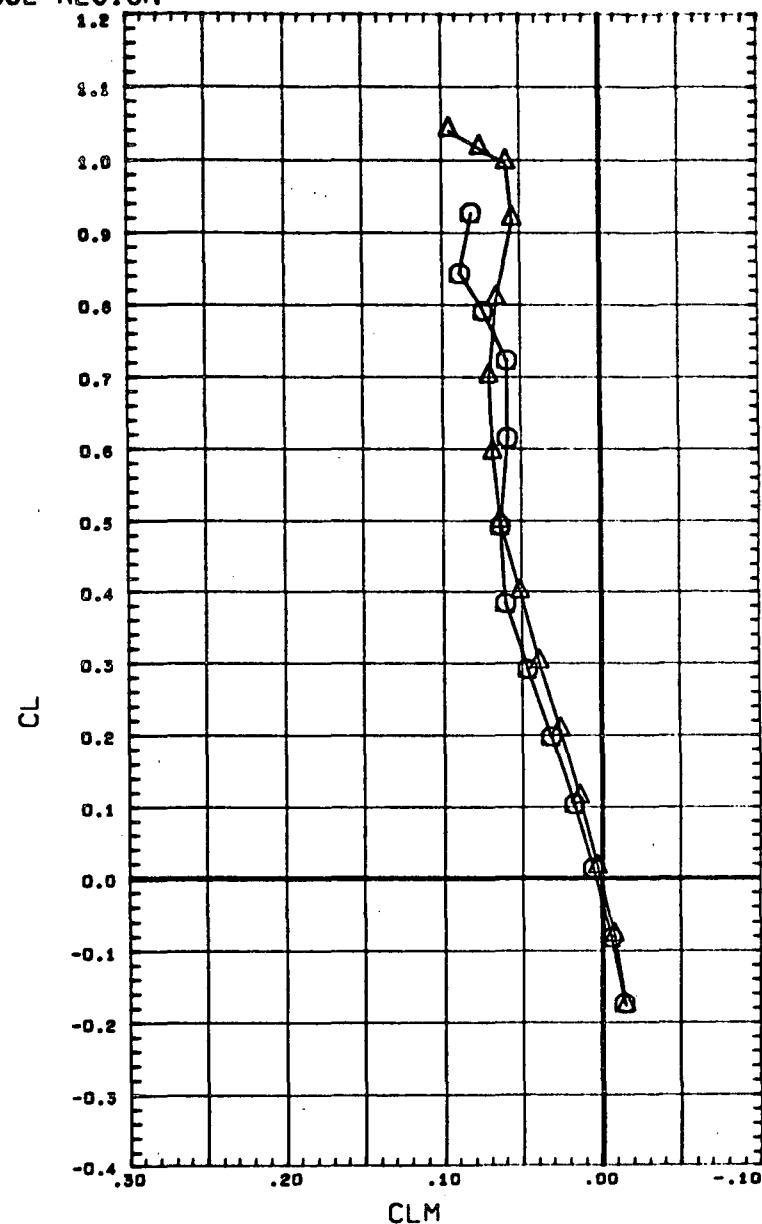
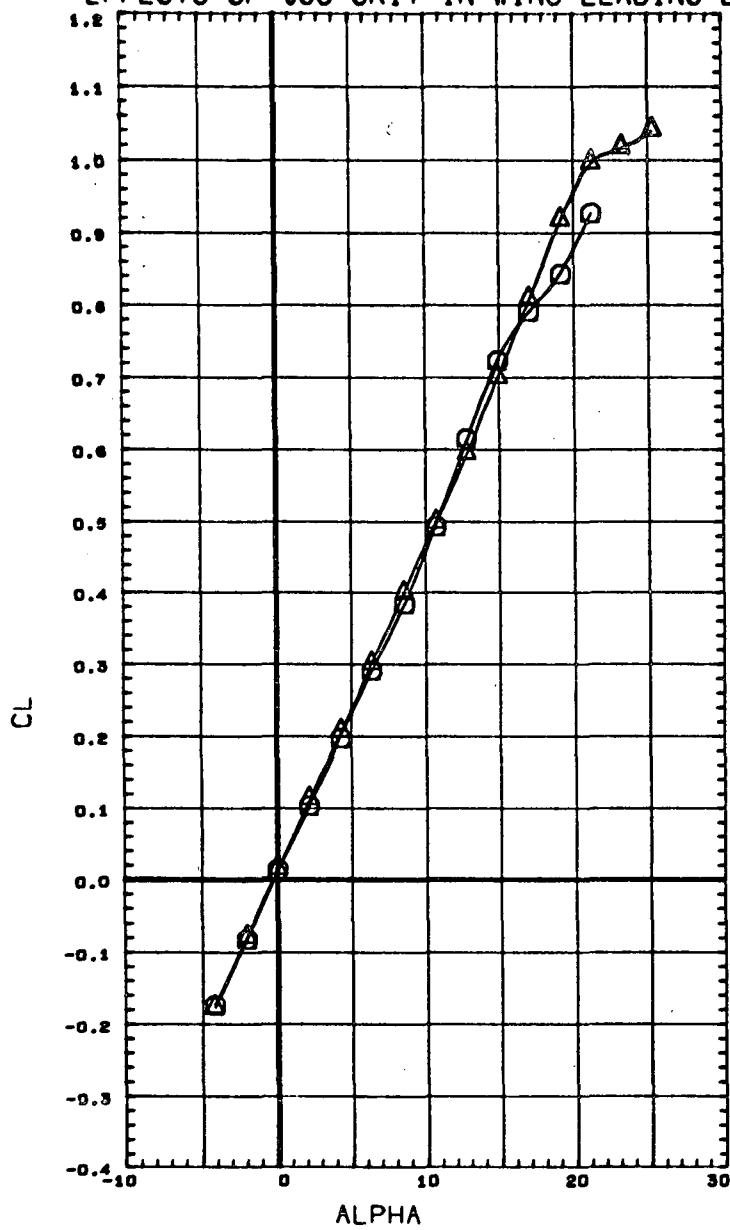
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6093) GDLST 603-0 B30  
 (AD6094) GDLST 603-0 B30  
 (AD6095) GDLST 603-0 B32  
 (AD6096) GDLST 603-0 B32

MACH 0.201

BETA CANARD  
 0.000  
 5.000  
 0.000  
 5.000

REFERENCE INFORMATION		
SREF	492.4804	SQ. IN.
LREF	16.1880	IN.
BREF	34.6320	IN.
XMRP	29.0780	IN.
YMRP	0.0000	IN.
ZMRP	0.0000	IN.
SCALE	0.0200	

## EFFECTS OF 036 GRIT IN WING LEADING EDGE REGION

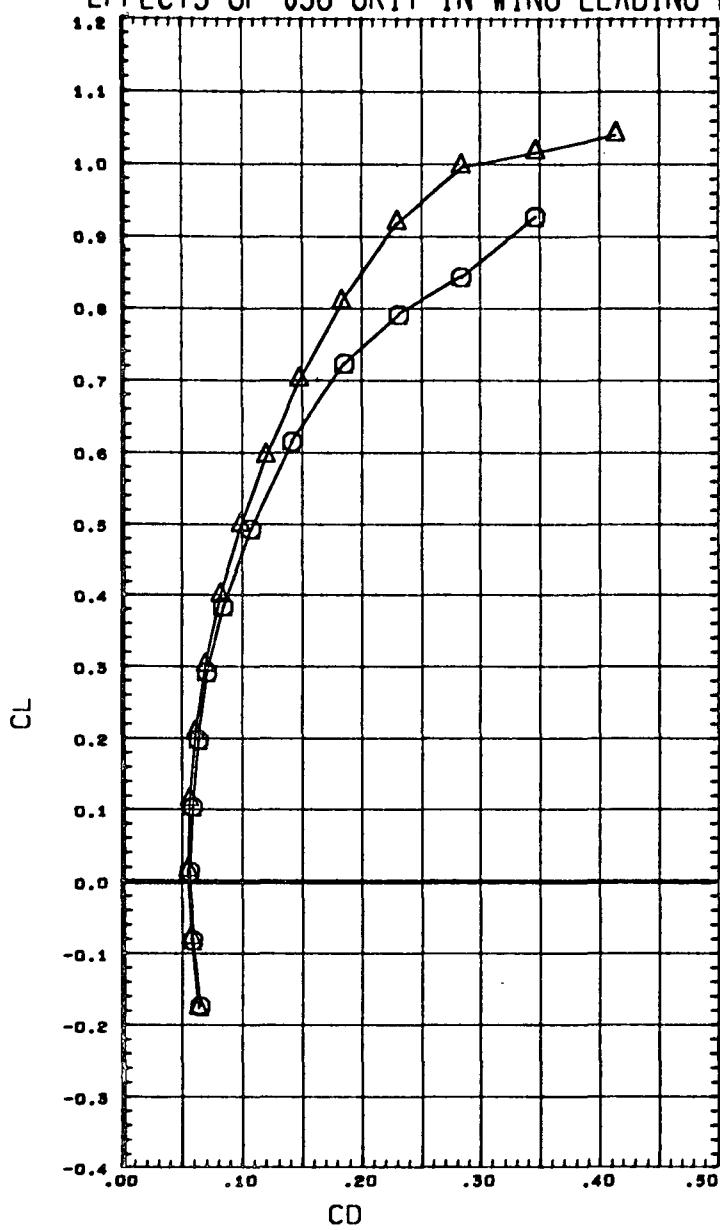


DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6042) Q GDLST 603-0 B30W23C10V14E38X7  
 (AD6002) R GDLST 603-0 B30W23C10V14E38

MACH 0.201

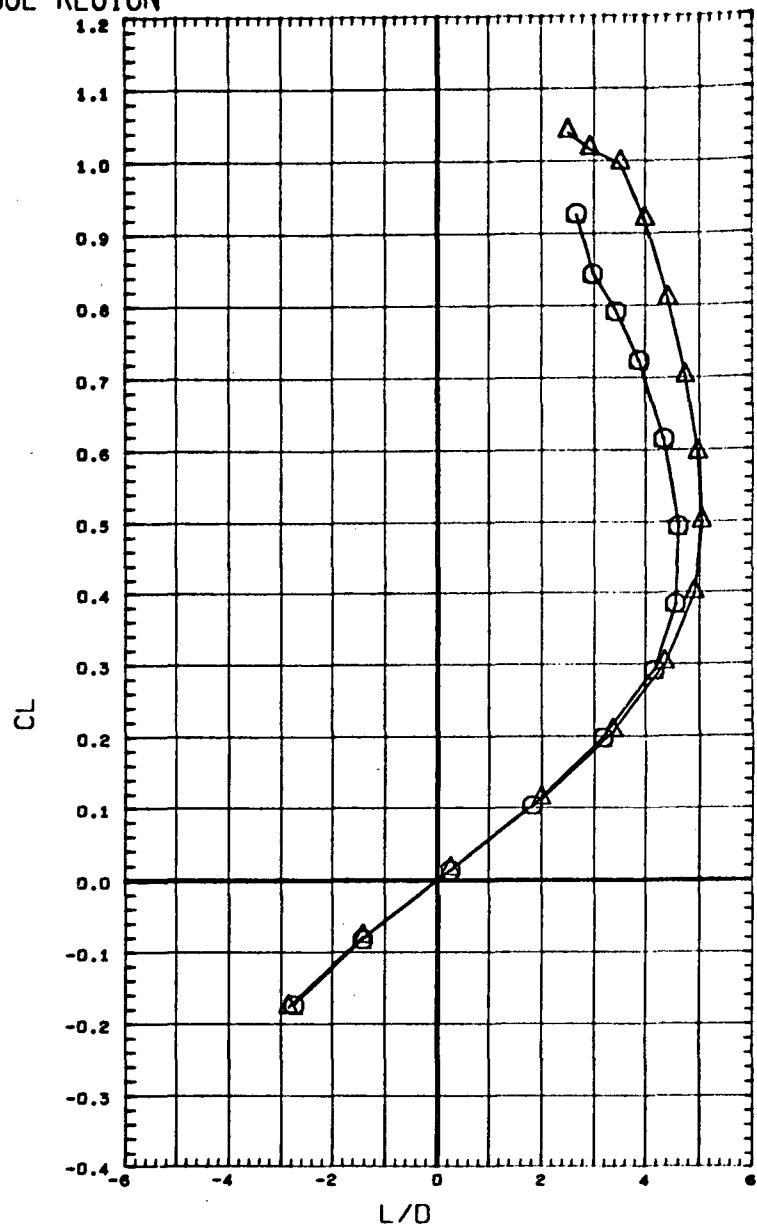
BETA	ELEVTR	CANARD	AILRON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN.
0.000	0.000	0.000	0.000	LREF	16.1880 IN.
				BREF	34.6320 IN.
				XMRP	29.0780 IN.
				YMRP	0.0000 IN.
				ZMRP	0.0000 IN.
				SCALE	0.0200

# EFFECTS OF 036 GRIT IN WING LEADING EDGE REGION



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6042) Q GDLST 603-0 B30W23C10V14E38X7  
 (AD6002) A GDLST 603-0 B30W23C10V14E38

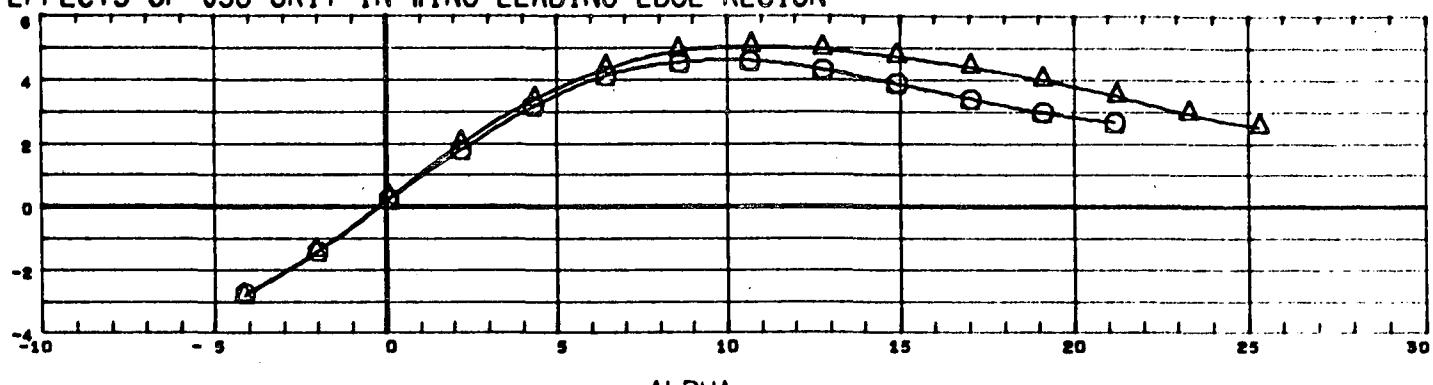
MACH 0.201



BETA ELEVTR CANARD AILRDN REFERENCE INFORMATION  
 0.000 0.000 0.000 0.000 SREF 492.4804 SQ. IN.  
 0.000 0.000 0.000 0.000 LREF 16.1880 IN.  
 BREF 34.6320 IN.  
 XMRP 29.0780 IN.  
 YMRP 0.0000 IN.  
 ZMRP 0.0000 IN.  
 SCALE 0.0200

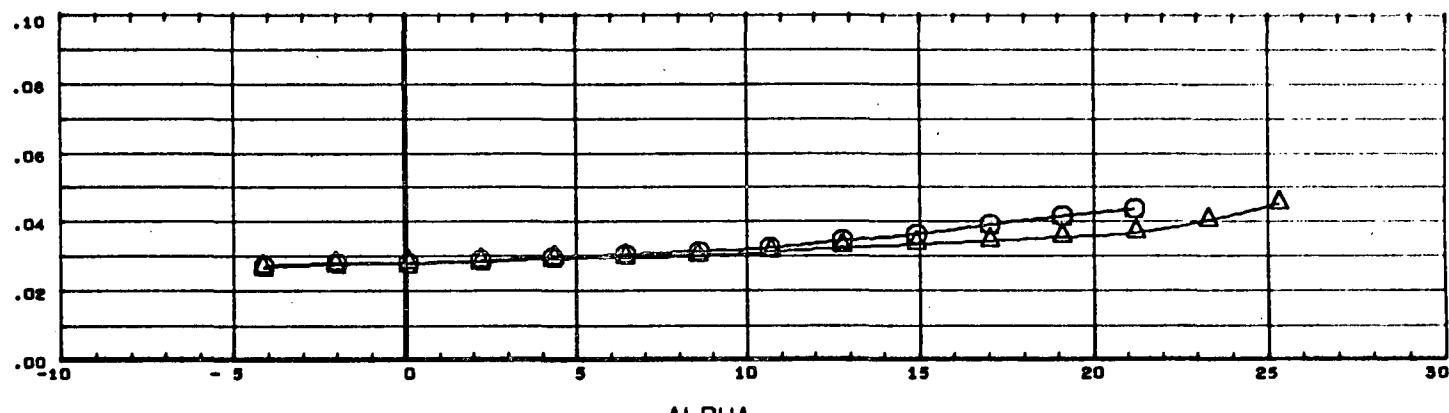
## EFFECTS OF 036 GRIT IN WING LEADING EDGE REGION

L/D



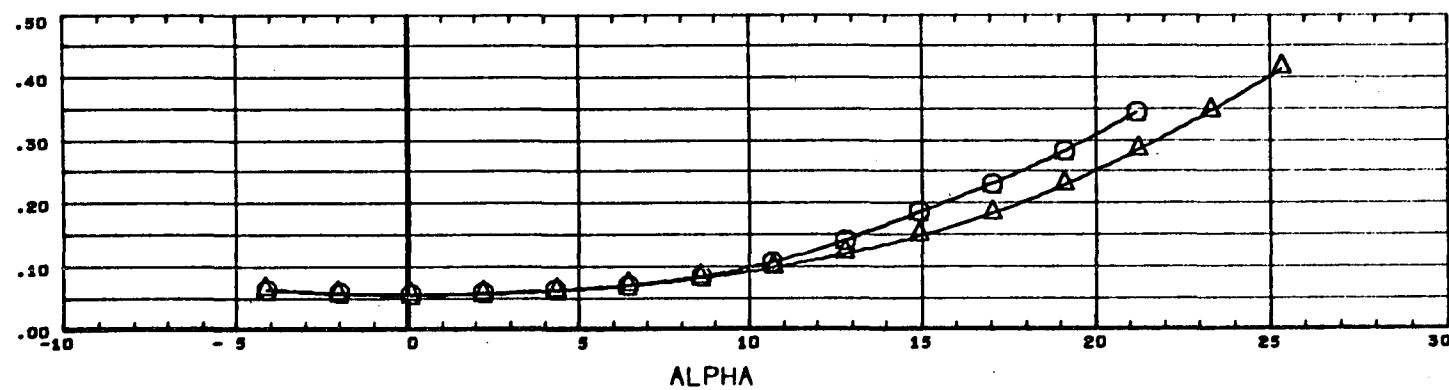
ALPHA

CAB



ALPHA

CD



ALPHA

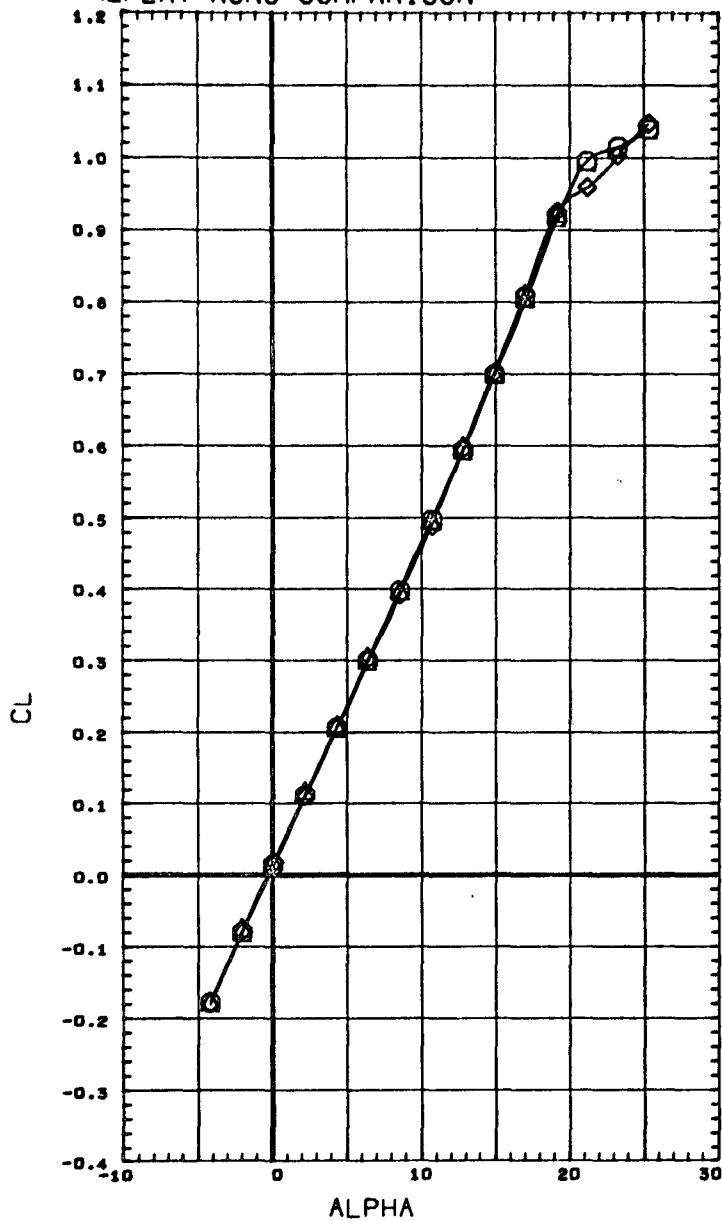
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6042) GDLST 603-0 B30W23C10V14E38X7  
 (AD6002) GDLST 603-0 B30W23C10V14E38

BETA	ELEVTR	CANARD	AIRLON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN.
0.000	0.000	0.000	0.000	LREF	16.1880 IN.
				BREF	34.6320 IN.
				XMRP	29.0780 IN.
				YMRP	0.0000 IN.
				ZMRP	0.0000 IN.
				SCALE	0.0200

MACH 0.201

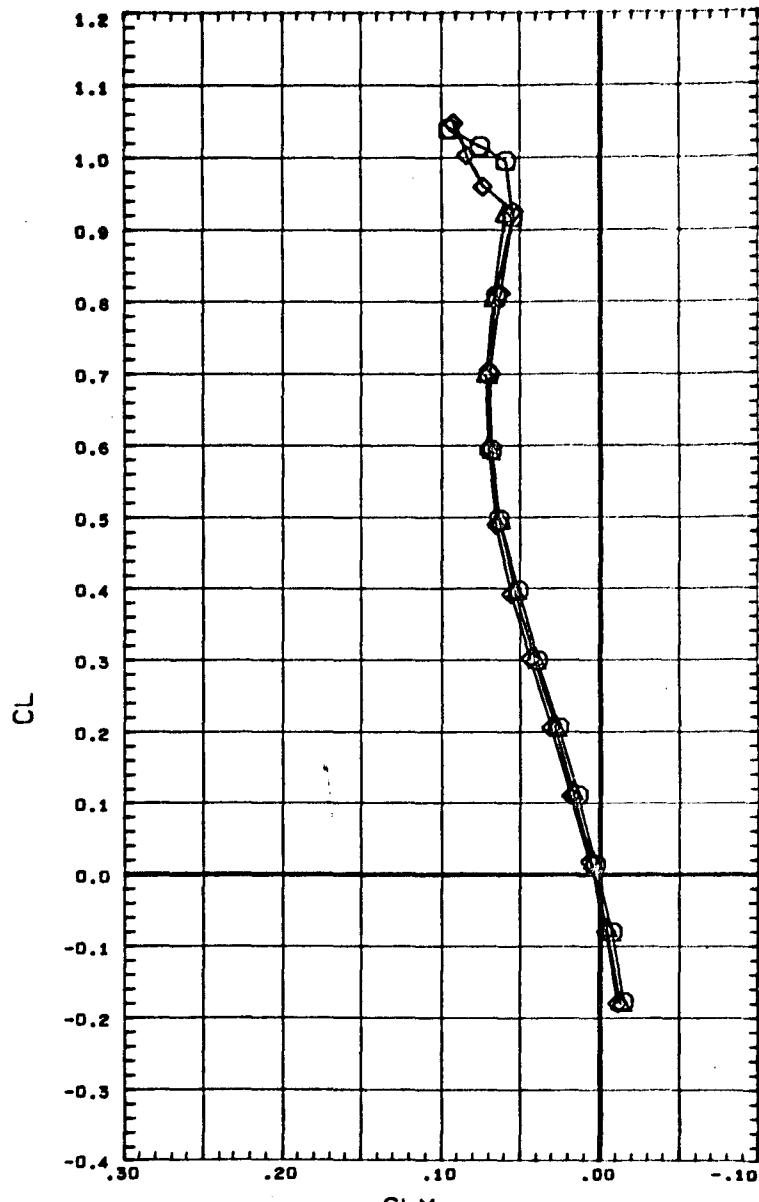
PAGE 66

## REPEAT RUNS COMPARISON



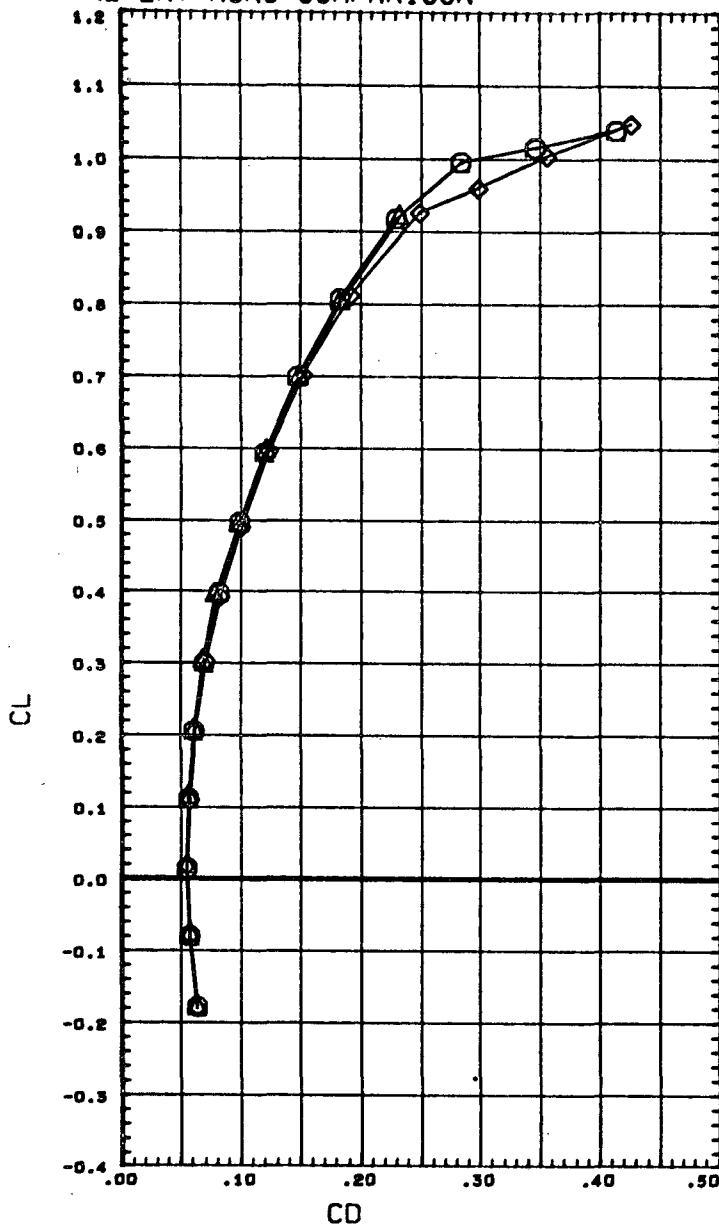
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (AD6002) Q GDLST 603-0 B30W23C10V14E38  
 (AD6043) O GDLST 603-0 B30W23C10V14E38  
 (AD6076) D GDLST 603-0 B30W23C10V14E38 (MODEL UPRIGHT)

MACH 0.201



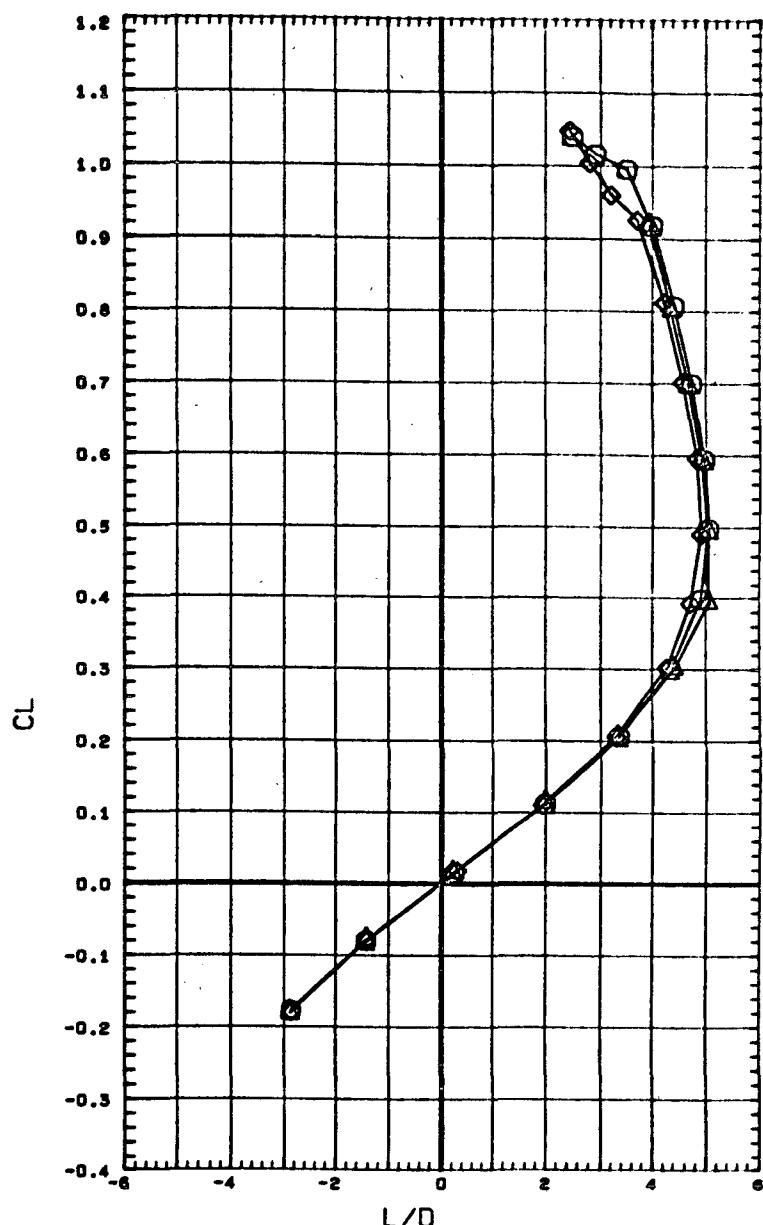
BETA ELEVTR CANARD AILRON REFERENCE INFORMATION  
 0.000 0.000 0.000 0.000 SREF 492.4804 SQ. IN.  
 0.000 0.000 0.000 0.000 LREF 16.1880 IN.  
 0.000 0.000 0.000 0.000 BREF 34.6320 IN.  
 XMRP 29.0760 IN.  
 YMRP 0.0000 IN.  
 ZMRP 0.0000 IN.  
 SCALE 0.0200

## REPEAT RUNS COMPARISON



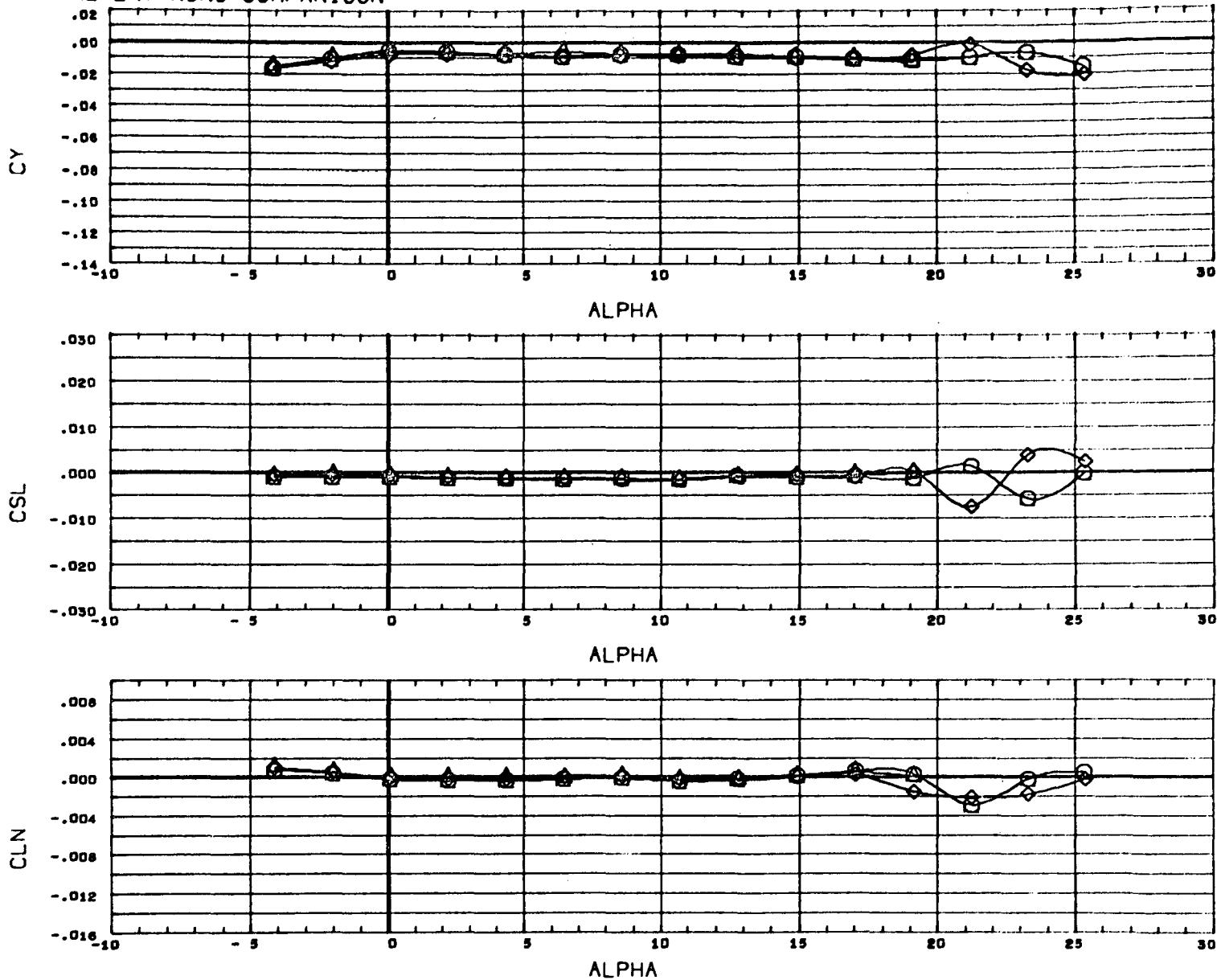
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(AD6002)	GDLST 603-0 B30W23C10V14E38
(AD6043)	GDLST 603-0 B30W23C10V14E38
(AD6076)	GDLST 603-0 B30W23C10V14E38 (MODEL UPRIGHT)

MACH 0.201



BETA	ELEVTR	CANARD	AILRON	REFERENCE	INFORMATION
0.000	0.000	0.000	0.000	SREF	492.4804 SQ. IN.
0.000	0.000	0.000	0.000	LREF	16.1680 IN.
0.000	0.000	0.000	0.000	BREF	34.6320 IN.
				XMRP	29.0760 IN.
				YMRP	0.0000 IN.
				ZMRP	0.0000 IN.
				SCALE	0.0200

## REPEAT RUNS COMPARISON



DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	RUDDER	AIRLON	REFERENCE	INFORMATION	
(AD6002)	Q	GDLST 603-0 B30W23C10V14E38	0.000	0.000	0.000	0.000	SREF	492.4804	SQ. IN
(AD6043)	Q	GDLST 603-0 B30W23C10V14E38	0.000	0.000	0.000	0.000	LREF	16.1880	IN.
(AD6076)	D	GDLST 603-0 B30W23C10V14E38 (MODEL UPRIGHT)	0.000	0.000	0.000	0.000	BREF	34.6320	IN.
							XMRP	29.0780	IN.
							YMRP	0.0000	IN.
							ZMRP	0.0000	IN.
							SCALE	0.0200	

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