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-SPACE SHUTTLE-



(NASA-CR-120045) HEAT TRANSFER
INVESTIGATION OF Langley RESEARCH CENTER
TRANSITION MODELS AT A MACH NUMBER OF 8,
VOLUME 2 R.K. Matthews, et al (Chrysler
Corp.) Mar. 1972 132 p |

HEAT TRANSFER INVESTIGATION OF Langley RESEARCH CENTER TRANSITION MODELS AT A MACH NUMBER OF 8

by

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25140

VKF 50-INCH
HYPERSONIC TUNNEL B

Arnold Engineering
Development Center

SADSAC SPACE SHUTTLE
AEROTHERMODYNAMIC
DATA MANAGEMENT SYSTEM

CONTRACT NAS8-4016
MARSHALL SPACE FLIGHT CENTER

SPACE DIVISION  CHRYSLER
CORPORATION

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March, 1972

SADSAC/SPACE SHUTTLE

WIND TUNNEL TEST DATA REPORT

CONFIGURATION: Langley Research Center Transition Models

TEST PURPOSE: To Determine the Heat Transfer Rate Distributions at a
Mach Number of 8.

TEST FACILITY: AEDC VKF 50-Inch Hypersonic Tunnel B

TESTING AGENCY: AEDC-MSFC

TEST NO. & DATE: VT 1162-5 June and September 1972

FACILITY COORDINATOR: L. L. Trimmer, ARO, INC.

PROJECT ENGINEER(S): R. K. Matthews, W. R. Martindale, ARO, INC.

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CONTRACT NAS 8-4016.

AMENDMENT 153

DRL 184 - 58

This report has been prepared by Chrysler Corporation Space Division under a Data Management Contract to the NASA. Chrysler assumes no responsibility for the data presented herein other than its display characteristics.

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FOREWORD

The work reported herein was sponsored by the Marshall Space Flight Center (MSFC), NASA. The results of tests presented were obtained by ARO, Inc. (a subsidiary of Sverdrup and Parcel and Associates, Inc.), contract operator of the Arnold Engineering Development Center (AEDC), AFSC, Arnold Air Force Station, Tennessee. Ascent and reentry conditions were simulated on shuttle models designed by McDonnell Douglas (MDAC), North American Rockwell (NAR) and General Dynamics Convair (GDC). In addition a limited amount of data were obtained on two research models provided by the Langley Research Center (LRC). Because of the broad scope of these tests the data will be presented in a series of SADSAC reports. This report presents the results of the phase-change paint test conducted at Mach 8 in Tunnel B on the Langley Research Center models.

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NOMENCLATURE

ALPHA-MODEL (α)	Model angle of attack, deg
ALPHA-PREBEND	Sting prebend angle, deg
ALPHA-SECTOR	Tunnel sector pitch angle, deg
H(T_0) or H	Heat-transfer coefficient based on $T_{aw} = T_0$, BTU/ ft^2 - sec - °R, and

$$H(T_0) = \frac{\beta \sqrt{\rho c k}}{\sqrt{\Delta t}}$$

where β is obtained from

$$\frac{T_{pc} - T_i}{T_{aw} - T_i} = 1 - e^{\beta^2} \operatorname{erfc}\beta$$

and $\Delta t \sim$ del time

T_{pc} ~ phase-change paint temperature, °R

T_i ~ initial model temperature, °R

T_{aw} ~ adiabatic wall temperature, °R

$\sqrt{\rho c k}$ ~ model material properties = $0.11 - 0.008\sqrt{\Delta t}$

or 0.037 BTU/ ft^2 -sec $^{1/2}$ - °R

$H(.9T_0)$ Heat transfer coefficient based on $T_{aw} = 0.9T_0$

$H(.85T_0)$ Heat transfer coefficient based on $T_{aw} = .85T_0$

HREF Reference heat transfer coefficient based on Fay-Riddell theory, BTU/ ft^2 -sec °R

$$H_{REF} = \left[\frac{8.139(P_{01})^{0.5} (\mu_{-0})^{0.4} (1 - P_{-INF}/P_{01})^{0.25}}{(R)^{0.5} (T_0)^{0.15}} \right] x$$

$$[0.2235 + 0.0000135 (T_0 + 760)]$$

where P_{01} ~ stagnation pressure downstream of a normal shock, psia

μ_{-0} ~ air viscosity based on T_0 , lb $_f$ sec/ ft^2

R ~ reference nose radius, (0.056 ft)

L	Model length (24.0 in.)
MU-INF	Free-stream viscosity, lb-sec/ft ²
P-INF	Free-stream static pressure, psia
P0	Tunnel stilling chamber pressure, psia
Q-INF	Free-stream dynamic pressure, psia
RE/FT	Free-stream unit Reynolds number, ft ⁻¹
RHO-INF	Free-stream density, slugs/ft ³
ROLL-MODEL	Model roll angle, deg
ST(T0)	Stanton number based on T ₀ ,
	$ST(T_0) = \frac{H(T_0)}{\rho_\infty V_\infty [0.2235 + 0.0000135 (T_0 + T_{pc})] \times (32.17)}$
STREF	Reference Stanton number
	$STREF = \frac{HREF}{\rho_\infty V_\infty [0.2235 + 0.0000135 (T_0 + T_{pc})] \times (32.17)}$
T-INF	Free-stream static temperature, °R
T0	Tunnel stilling chamber temperature, °R
TW	Model wall temperature, °R
TIME	Time from start of model injection, sec
DEL-TIME (Δt)	Time model exposed to airstream, sec
V-INF (V_∞)	Free-stream velocity, ft/sec
YAW	Model yaw angle, deg
X,Y,Z	Model coordinates (see Fig. 1), in.
YMAX	Local semi-span at a given model station, in.

SECTION 1

INTRODUCTION

This report presents the results of a wind tunnel test program to determine aerodynamic heat transfer distributions on the Langley Research Center transition models. The tests were conducted at the Arnold Engineering Development Center (AEDC) in Tunnel B of the von Karman Gas Dynamics Facility (VKF). The test period was in June and September 1971.

Heat-transfer rates were determined by the phase-change paint technique on Stycast[®] and RTV models using Tempilaq[®] as the surface temperature indicator. The nominal test conditions were: Mach 8, length Reynolds numbers of 5.0×10^6 and 7.4×10^6 , and angles of attack of 20, 40, and 60 deg.

Model details, test conditions, and reduced heat-transfer data are presented in this report. Data reduction of the phase-change paint photographs was performed by VKF personnel utilizing a new technique which is described in the data presentation section.

SECTION 2

MODELS AND APPARATUS

2.1 MODEL DESCRIPTION

Drawings and photographs of the models are presented in Figures 1 and 2 respectively. The models were provided by the Langley Research Center and were fabricated with a 1/4-in. layer of Stycast over a fiberglass mandrel. During the June entry the Stycast cracked on the windward surface of the delta body and as a result the Stycast layer was removed and replaced with RTV prior to the September entry.

2.2 FACILITY DESCRIPTION

Tunnel B is a continuous, closed-circuit, variable density wind tunnel with an axisymmetric contoured nozzle and a 50-in.-diam test section. The tunnel can be operated at a nominal Mach number of 6 or 8 at stagnation pressures from 20 to 300 and 50 and 900 psia, respectively, at stagnation temperatures up to 1350°R. The model may be injected into the tunnel for a test run and then retracted for model cooling or model changes without interrupting the tunnel flow.

SECTION 3

PROCEDURES

3.1 TEST TECHNIQUE

Prior to each run the model was cleaned and cooled with alcohol and then spray painted with Tempilaq. The models were installed on the model injection mechanism at the desired test attitude and the model temperature was measured with a thermocouple probe or with model-embedded thermocouples. During the course of the test many of the embedded thermocouples became inoperative and the probe temperature was generally used to determine the model initial temperature. The model was then injected into the airstream for approximately 15 seconds and during this time the model surface temperature rise produced isotherm melt lines. The progression of the melt lines was photographed with 70-mm sequenced cameras operating at two frames per second.

3.2 TEST CONDITIONS

Nominal test conditions are presented in the data summary sheets (Table 1). The specific test conditions for each run (or group) are provided on the data tabulation sheets preceding each set of melt line tracings. As mentioned in the foreword this test was part of a comprehensive Space Shuttle investigation and as a result the run numbers are not consecutive.

3.3 DATA REDUCTION

During each run the tunnel conditions and time of each picture were recorded on magnetic tape. The heat transfer coefficient for each

picture was calculated from the semi-infinite slab transient heat conduction equation.

$$\frac{T_{pc} - T_i}{T_{aw} - T_i} = 1 - e^{\beta^2} \operatorname{erfc} \beta$$

where $\beta = \frac{h\sqrt{\Delta t}}{\sqrt{\rho c k}}$, and $\sqrt{\rho c k} = 0.11 - 0.008 \sqrt{\Delta t}$ for StyCast, $\sqrt{\rho c k} = 0.037$ for RTV.

The equation for the thermal properties ($\sqrt{\rho c k}$) of StyCast was obtained by evaluation of a considerable amount of hemisphere calibration data and supplemented by VKF laboratory measurements. The value of $\sqrt{\rho c k}$ for RTV was obtained from Langley personnel.

Heat-transfer coefficients were calculated from assumed adiabatic wall temperatures of T_0 , $0.9T_0$, and $0.85T_0$ (see tabulated data sheets). The use of three values of T_{aw} provides an indication of the sensitivity of the heat-transfer coefficient (h) to the values of T_{aw} assumed. For the sake of consistency all plots and melt lines in this report are based on $T_{aw} = T_0$.

All heat-transfer coefficients were non-dimensionalized by dividing by the stagnation point heat-transfer coefficient (Ref. 1) on a 0.056-ft radius sphere.

SECTION 4

DATA PRESENTATION

The test results are presented as individual melt contours in body coordinates grouped as follows:

<u>Model</u>	<u>Re/ft</u>	<u>α, deg</u>
LRC-DB Bottom Surface	2.5×10^6	20,40,60
" " "	3.7×10^6	20,40,60
LRC-SB Bottom Surface	2.5×10^6	20,40,60
" " "	3.7×10^6	20,40,60

Preceding each set of melt contours is a tabulated data sheet which lists the specific test conditions and the time of each picture with the corresponding heat-transfer parameters. Following each set of surface contour plots are axial and spanwise data plots.

These figures were machine generated utilizing a data reduction technique recently developed at the VKF. This technique has simplified the tedious film reading task and provided body coordinates of melt line contours on magnetic tape.

The 70-mm film was projected onto an 8 x 10-in. glass plate and an experienced engineer traced the melt line contours. In regions of relatively constant heating a distinct melt line is extremely difficult to define and in some cases the melt line tracings were terminated because of poor definition. Of course, the melt line tracings are in picture plane coordinates, whereas body coordinates are desired. The transformation to body coordinates was accomplished as follows:

- 1) The model coordinates were measured at many model stations with a modified Sheffield Cordax coordinate measuring machine (Model 200),
- 2) the camera location relative to the model was determined,
- 3) using the principles of photogrammetry and the information obtained in steps 1 and 2, the model coordinates were transformed into the film plane,

4) the body coordinates of a given melt line were then obtained by interpolation in the film plane, with the results being stored on magnetic tape.

The level of the heat-transfer coefficient associated with each melt line was obtained by the procedure outlined in Section 3.3.

Table 2, Page 13, presents a summary of the plotted data.

REFERENCES

1. Fay, J. A. and Riddell, F. R. "Theory of Stagnation Point Heat Transfer in Dissociated Air." Journal of the Aeronautical Sciences, Vol. 25, 1958, pp. 73-85.

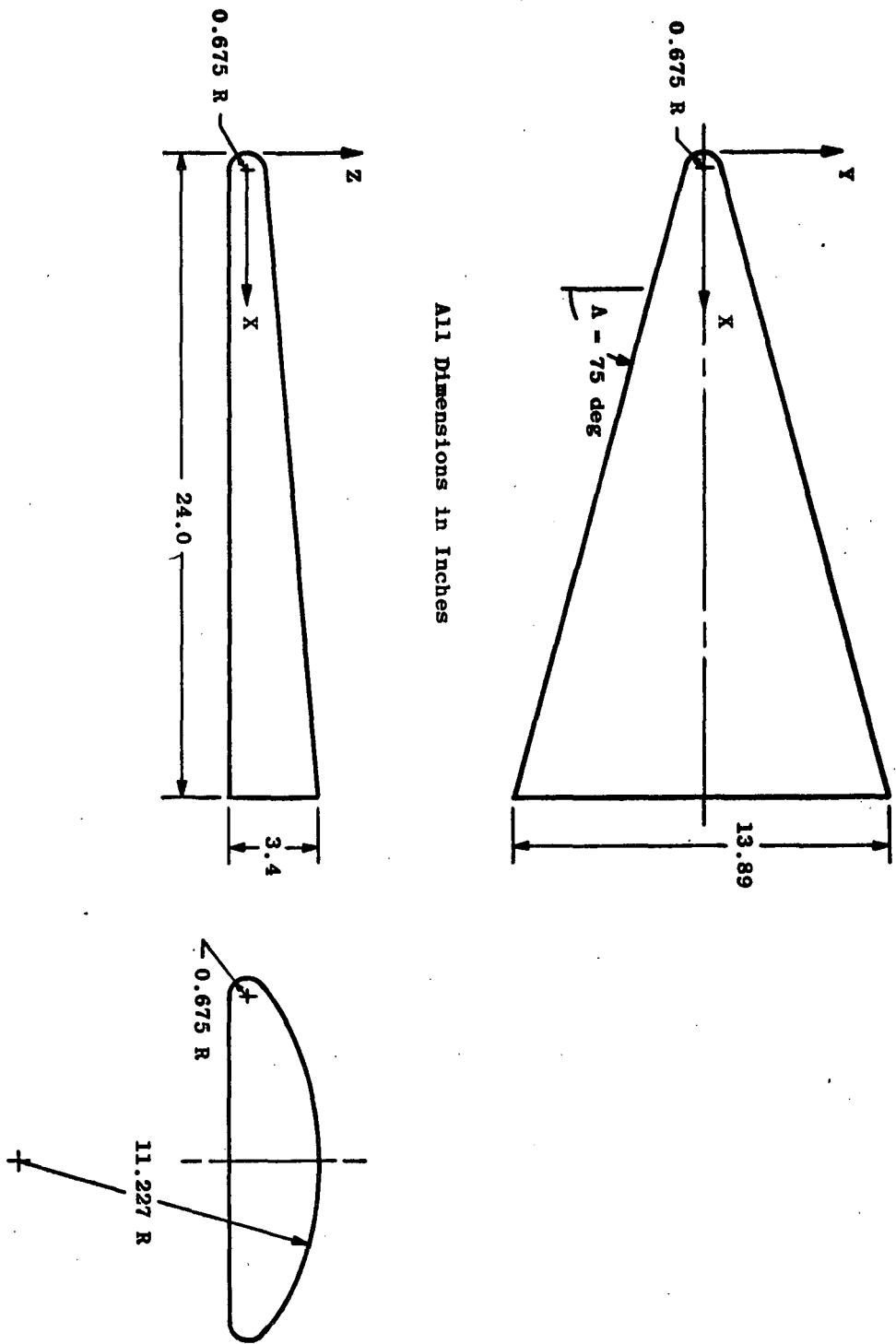
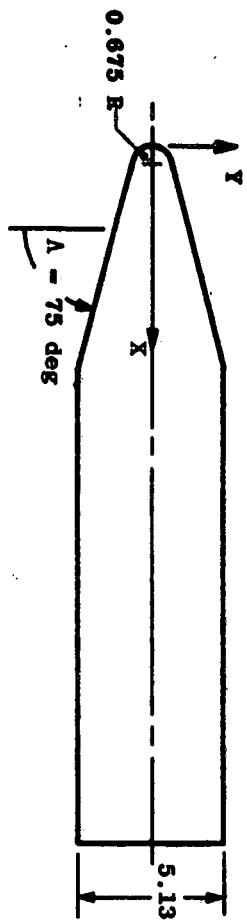
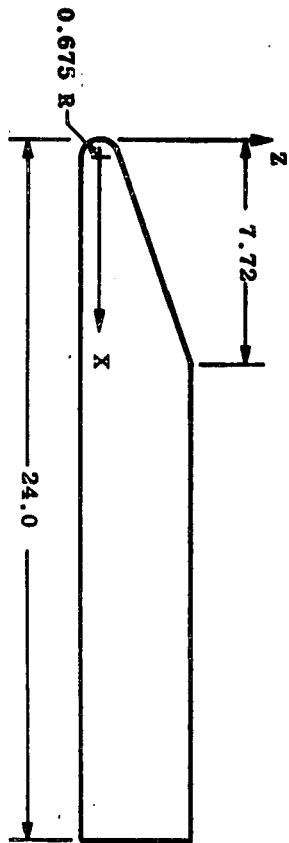


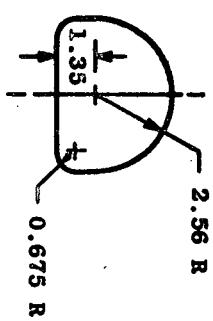
Fig. 1 a. Configuration 11, Delta Body (LRC-DB)
Sketches of the Langley Research Center Transition Models



All Dimensions in Inches



b. Configuration 12, Straight Body (LRC-SB)
Fig. 1 Concluded





a. Configuration 11, Delta Body (LRC-DB)

Fig. 2 Photographs of the Langley Research Center Transition Models



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b. Configuration 12, Straight Body (LRC-SB)

Fig. 2 Concluded

Table 1
PHASE CHANGE COATING TEST DATA SUMMARY SHEET

TEST TITLE: Phase-Change Coating Test on Langley Transition Models

TEST NUMBER: VT1162-5 **TEST FACILITY:** VKF Tunnel B

TEST DATE: June and Sept. 1971 **TEST ENGINEER:** R. K. Matthews

Run No.	Model Configuration Identification	Model Length in.	Free Stream Mach Number	Total Pressure (psia)	Total Temp. °R	T _{aw} * / Total Temp.	RNX 10 ⁶ Ft	Phase Change Temp. °F	Model Position (degrees)	Model Surface
147	LRC-DB (Conf. 11)	24.0	8.0	555	1310	1.0	2.5	113	20 0	180
129								200	40	
354								350	40	
360								350	60	
149								350	60	
163								200	20	
								200	20	
171								200	40	
343								400	40	
350								400	60	
								60	10	
143	LRC-SB (Conf. 12)			555	1310	2.5	125	20 0	180	
353								250	40	
136								200	60	
								60	10	

* T_{aw} = adiabatic wall temperature

Table 1 - concluded

PHASE CHANGE COATING TEST DATA SUMMARY SHEET

TEST TITLE: Phase-Change Coating Test on Langley Transition Models

TEST NUMBER: VT1162-5 TEST FACILITY: VKF Tunnel B
TEST DATE: June and Sept. 1971 TEST ENGINEER: R. K. Matthews

* T_{sw} = adiabatic wall temperature

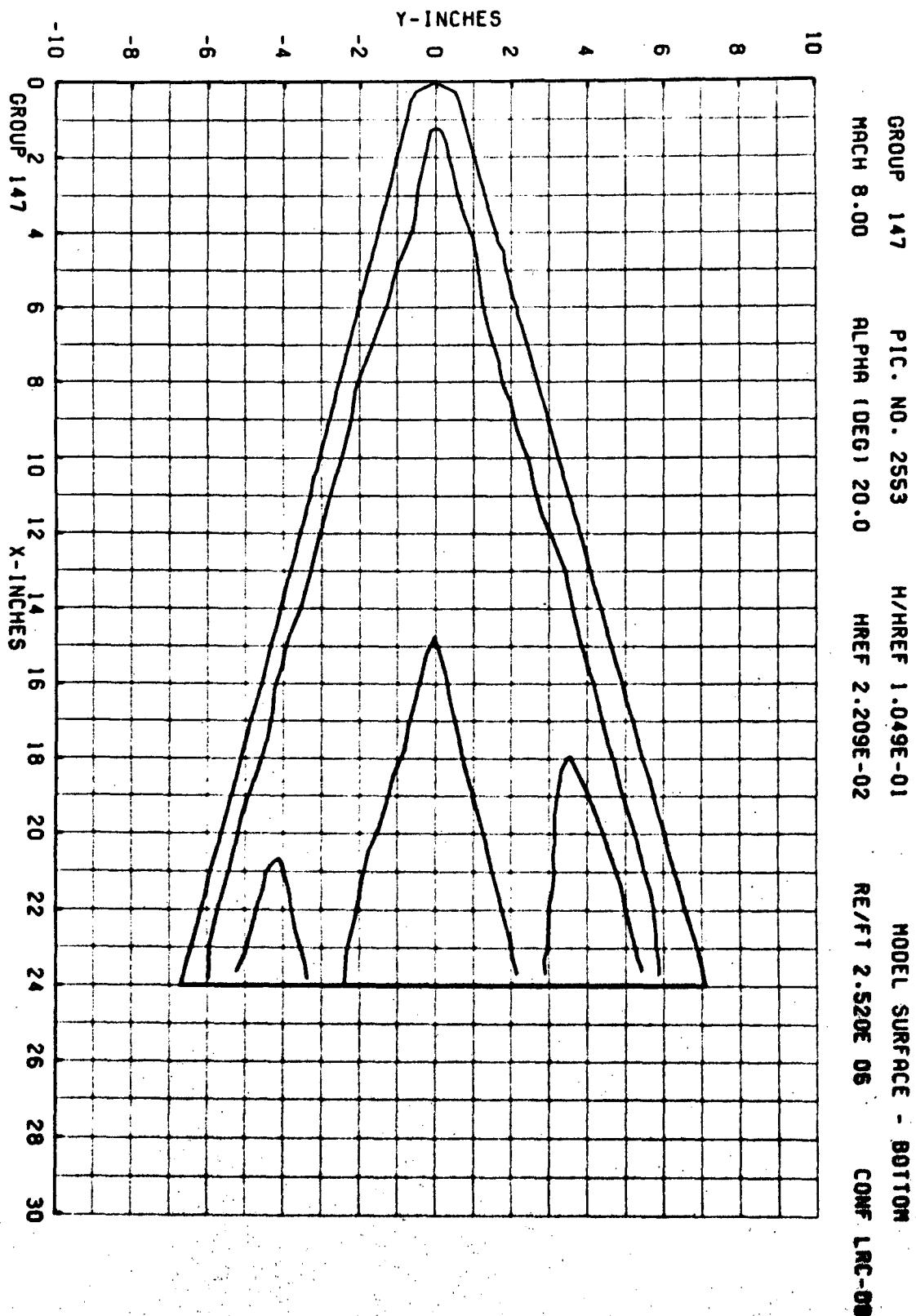
TABLE 2
SUMMARY DATA PLOT INDEX

MODEL CONFIGURATION	MODEL SURFACE	PAGES	Re/FT 10^6	ANGLE OF ATTACK - DEGREES
IRC-DB(CONF.11)	Bottom	15 - 17	X	20
		19 - 23	X	40
		25 - 32	X	60
		33 - 41	X	
		43 - 45	X	
		47 - 51	X	
		53 - 55	X	
		57 - 65	X	
IRC-DB(CONF.11)	Bottom	67 - 74	X	X
IRC-SB(CONF.12)	Bottom	76 - 80	X	X
		82 - 88	X	X
		90 - 95	X	
		97 - 100	X	
		102 - 106	X	
		108 - 113	X	
		115 - 120	X	
IRC-SB(CONF.12)	Bottom	122 - 126	X	X

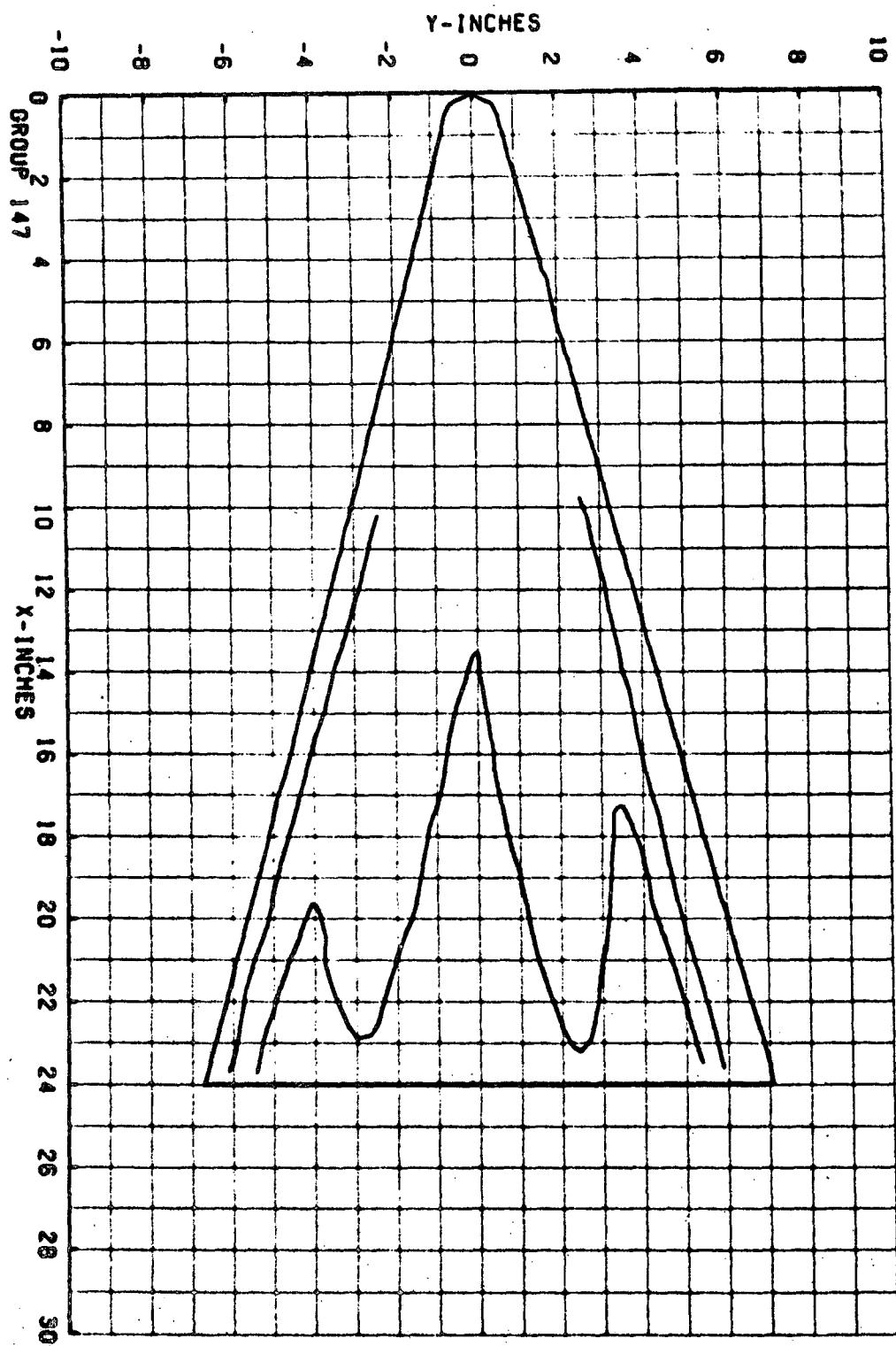
11 / 19

AEDC (AHO, INC.) ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL B.

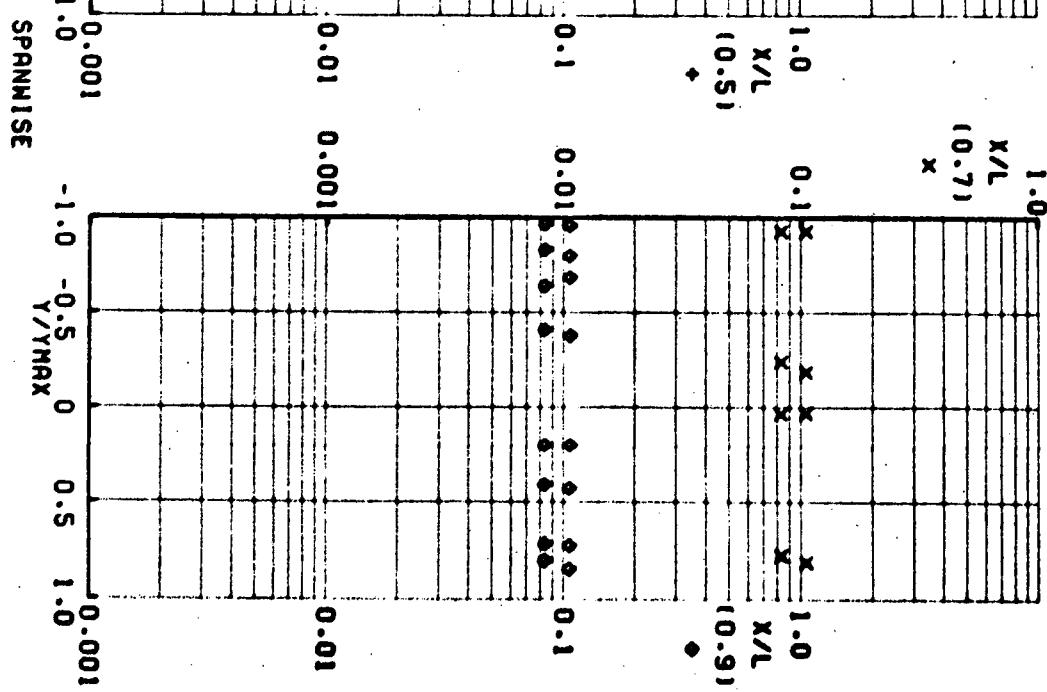
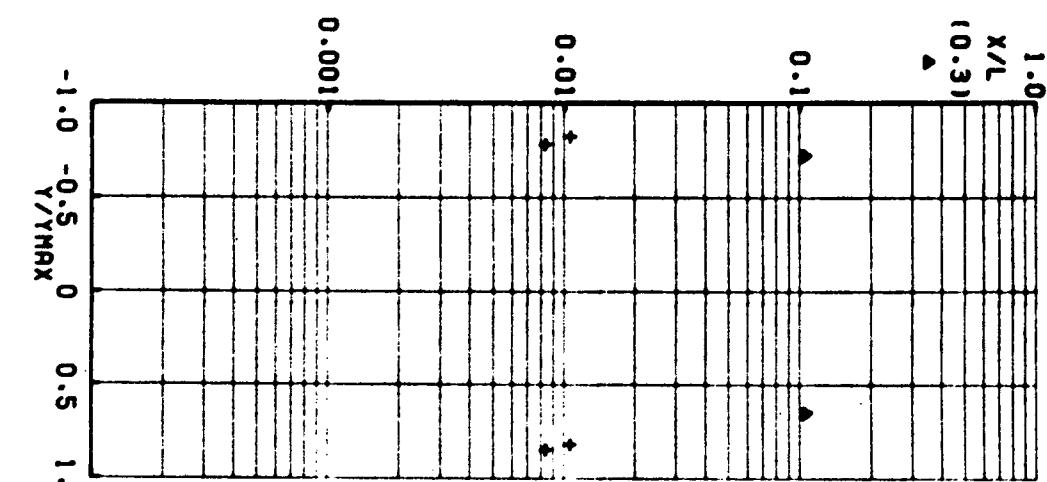
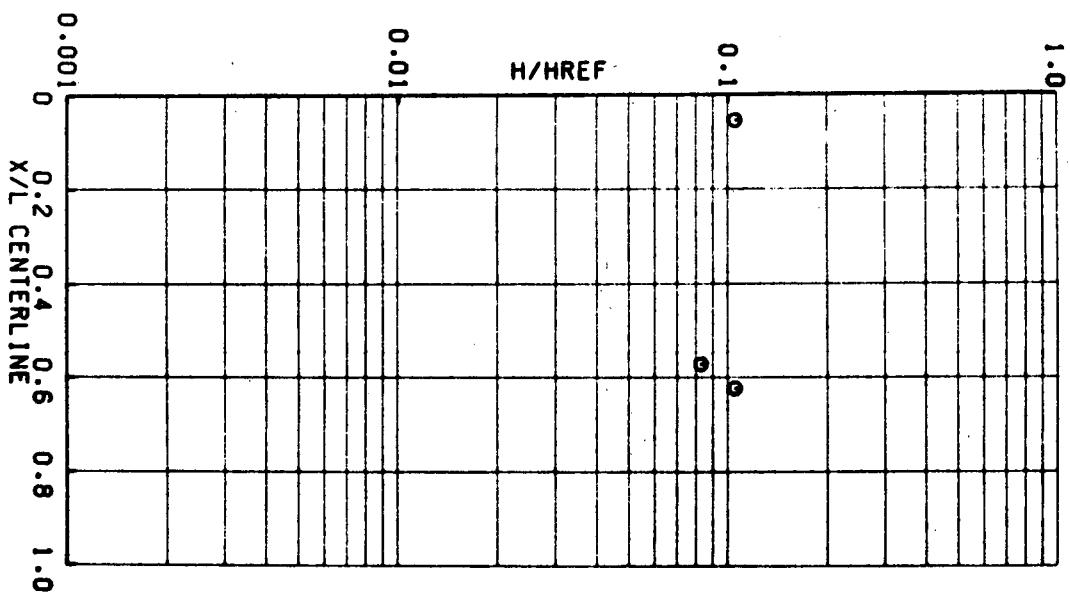
GROUP CONFIG-MODEL MACH-NR PUPSLA TO DEG R ALPHA-MODEL ALPHA-SECTOR ALPHA-PHEBEND ROLL-MODEL VAW
 147 11 LKC-DB 8.00 554.1 1305 20.00 3.00 -23.00 180.00 .0
 T-INF P-INF Q-INF V-INF RHO-INF MU-INF REF/T HREF STREF
 (DEG R) (PSIA) (PSIA) (F/SEC) (SLUGS/FT3) (LB-SEC/FT2) (FT-1) (R= .056FT) (K= .056FT)
 94.5 .057 2.543 3612 5.037E-05 7.012E-08 2.52E 06 2.209E-02 1.433E-02
 CAMERA PAINT TEMP (DEG F) INITIAL TEMP (DEG F) SQUARE ROOT (MMUXCK)
 TOP(T) 113 113 AVERAGE. IN = .77 -0.008(SQUARE MOUT DEL. TIME). 4.011
 SIDE(S) BOTTOM(B) 113



GROUP 147 PIC. NO. 2556 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 20.0 H/HREF 8.260E-02
HREF 2.209E-02 RE/FT 2.520E 06 CONF LRC-08



GROUP 147 ALPHA (DEG) 20.0 HREF 2.209E-02 MACH 8.00
MODEL SURFACE - BOTTOM RE/FT 2.520E 06 CONF LRC-08



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AFUDCA (AHO, INC.) ARNOLD AFS, TENNESSEE

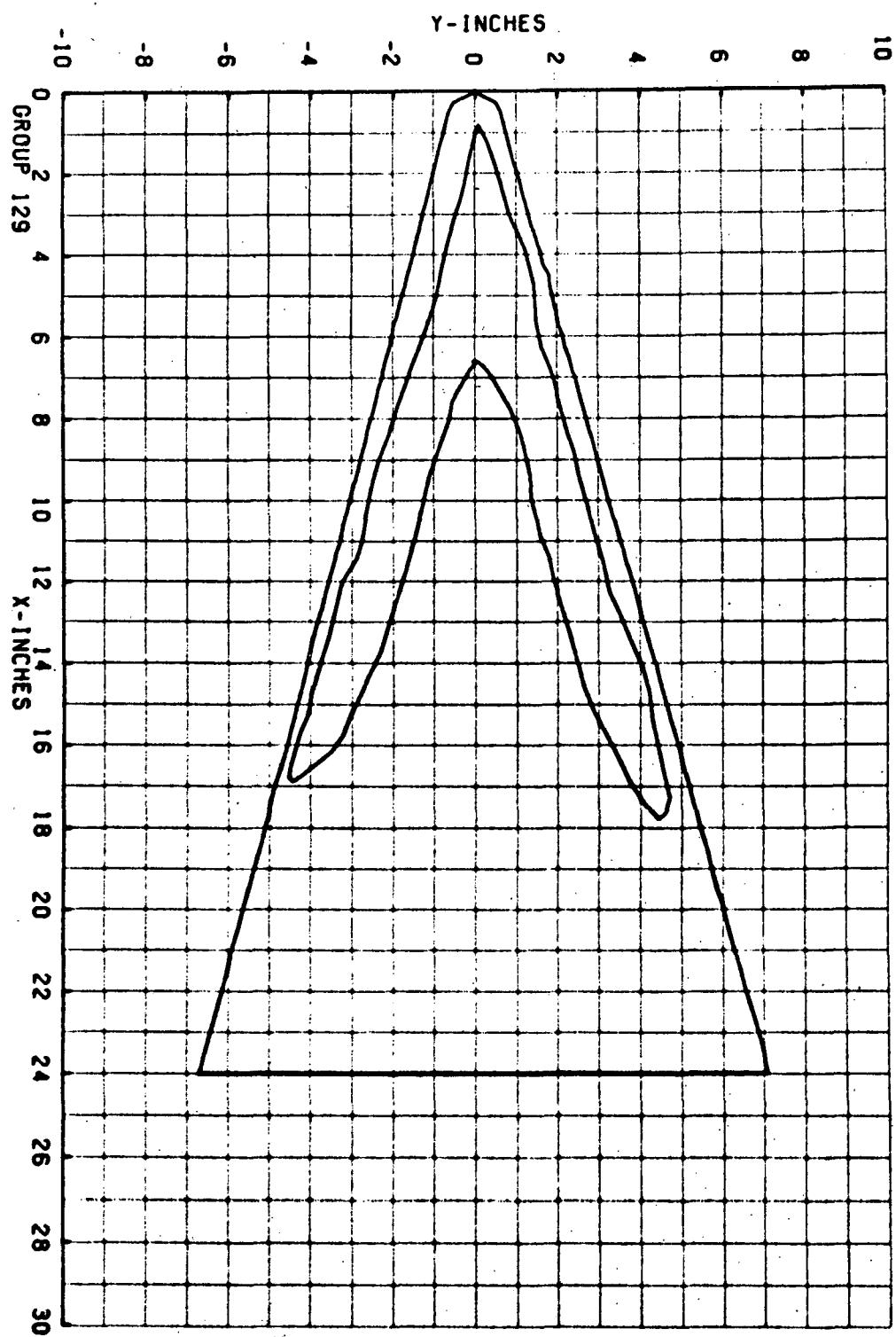
VON KARMAN GAS D
50 INCH HYPER

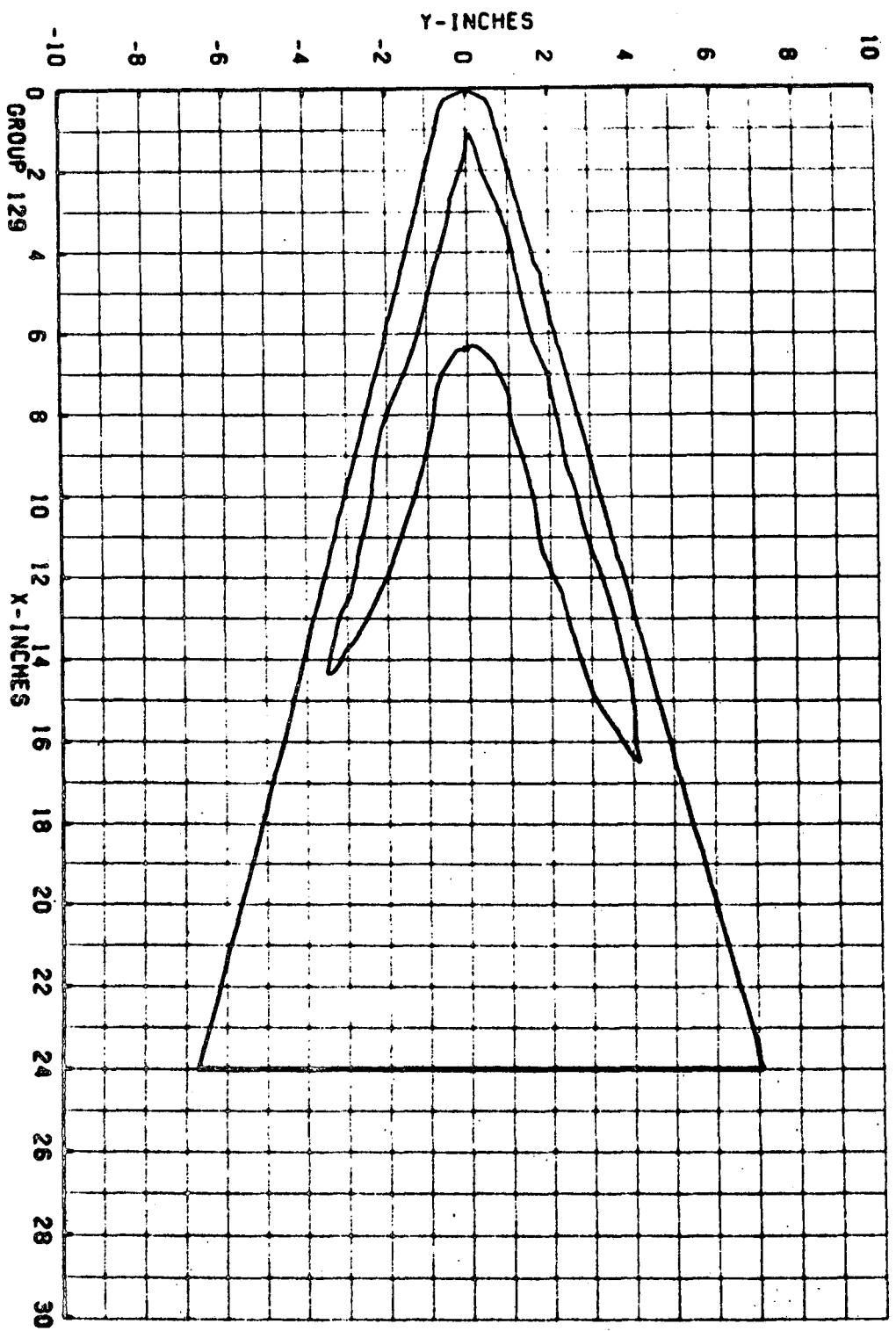
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    GROUP  CONFIG  MODEL   MACH NO  PROPSIA  TO DEG R  ALPHA-MODEL  ALPHA-SECTOR  ROLL-MODEL  yaw
    129    11  LCG-UH    8.00   551.9    1314    39.98    10.02   -50.00   180.00   .0
    T-INF  P-INF  J-INF  V-INF  RHO-INF  MU-INF  RE/FI  HREF  SIEF
    (DEG R) (PFTA) (JSTA) (VFTA) (SLUGS/FT3) (LB-SEC/FT2) (FT-1) (HE = .056FT) (HE = .056FT)
    95.3   .057   2.533   3826   4.9E-05   7.009E-09   2.48E-06   2.208E-02   1.442E-02
    CAMERA PAINT ITEMP (DEG F) INITIAL TEMP (DEG F) SQUARE ROOT (RHOXCK)
    TUP(1) 200     113      AVERAGE TM = 107. -0.008(SQUARE ROOT DEL TIME) + 0.11
    SIDE(S) 113
    BOTTOM(B)

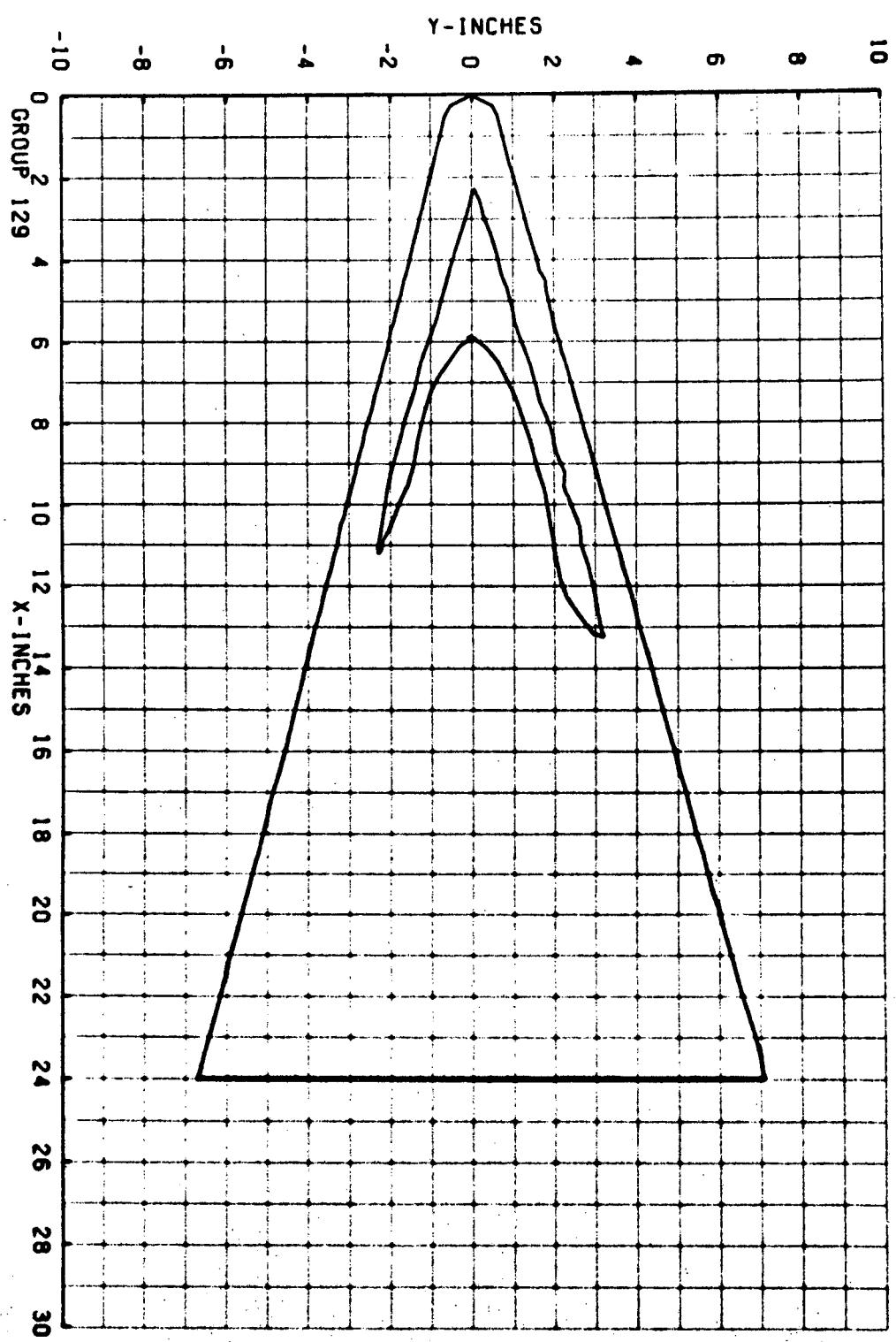
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GROUP 129 PIC. NO. 1949 MODEL SURFACE - BOTTOM
MACH 8.00 H/HREF 3.311E-01 HREF 2.208E-02 RE/FT 2.480E 06 CONF LRC-06
ALPHA (DEG) 40.0

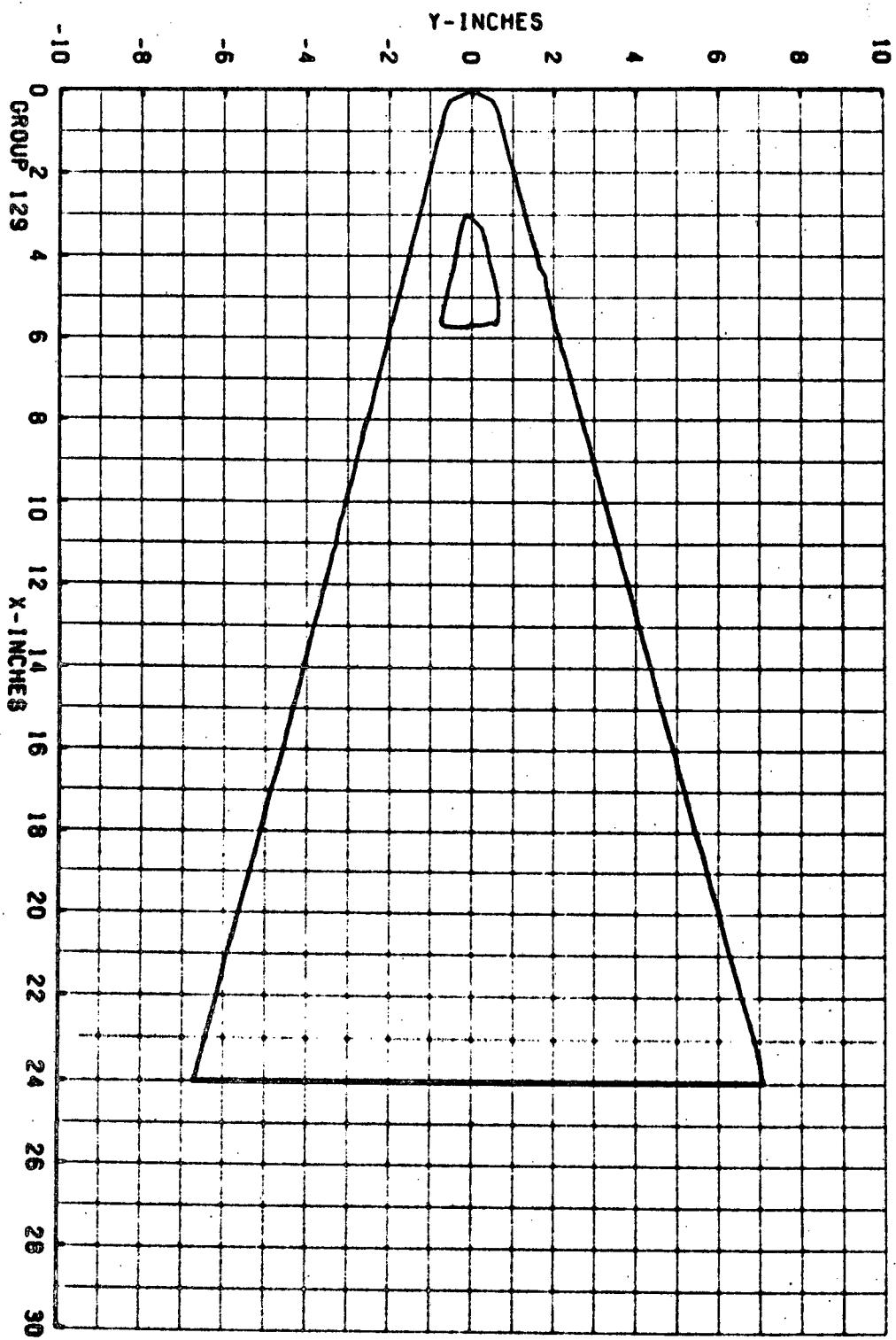




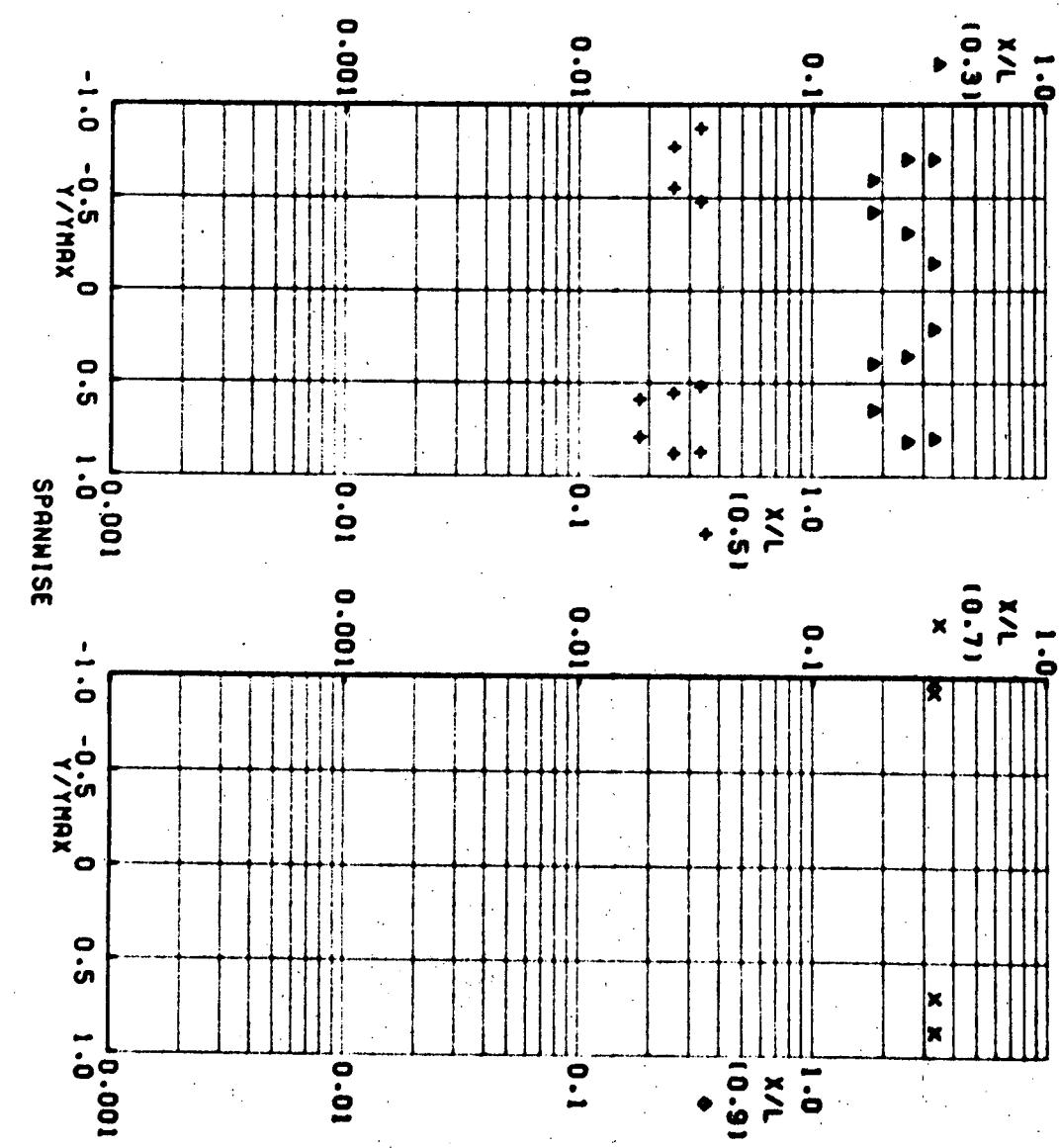
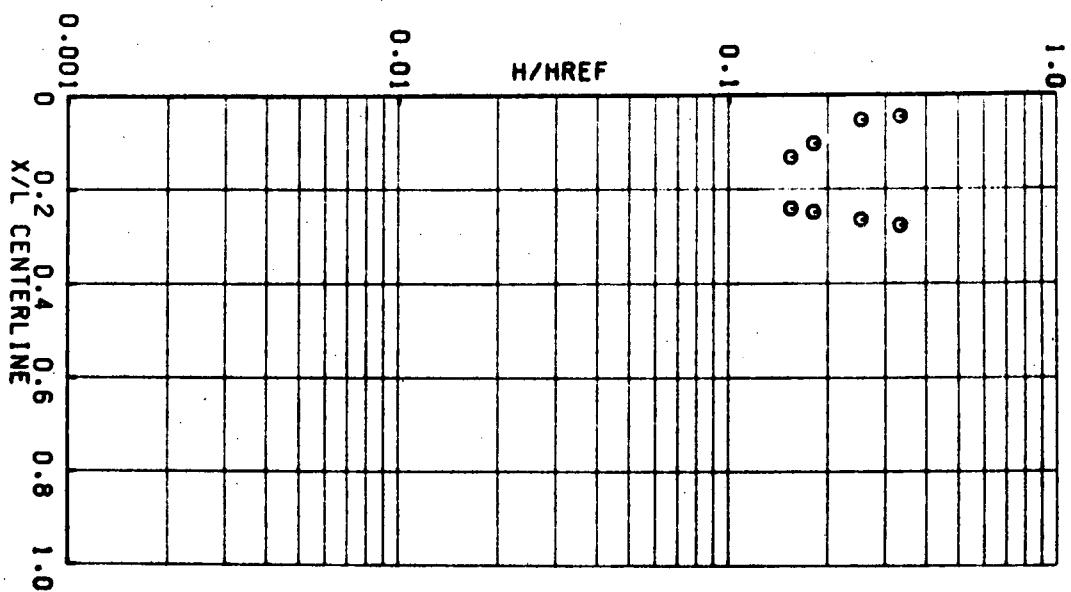
GROUP 129 PIC. NO. 1958 H/HREF 1.806E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.208E-02 RE/FI 2.480E 06 CONF LRC-08



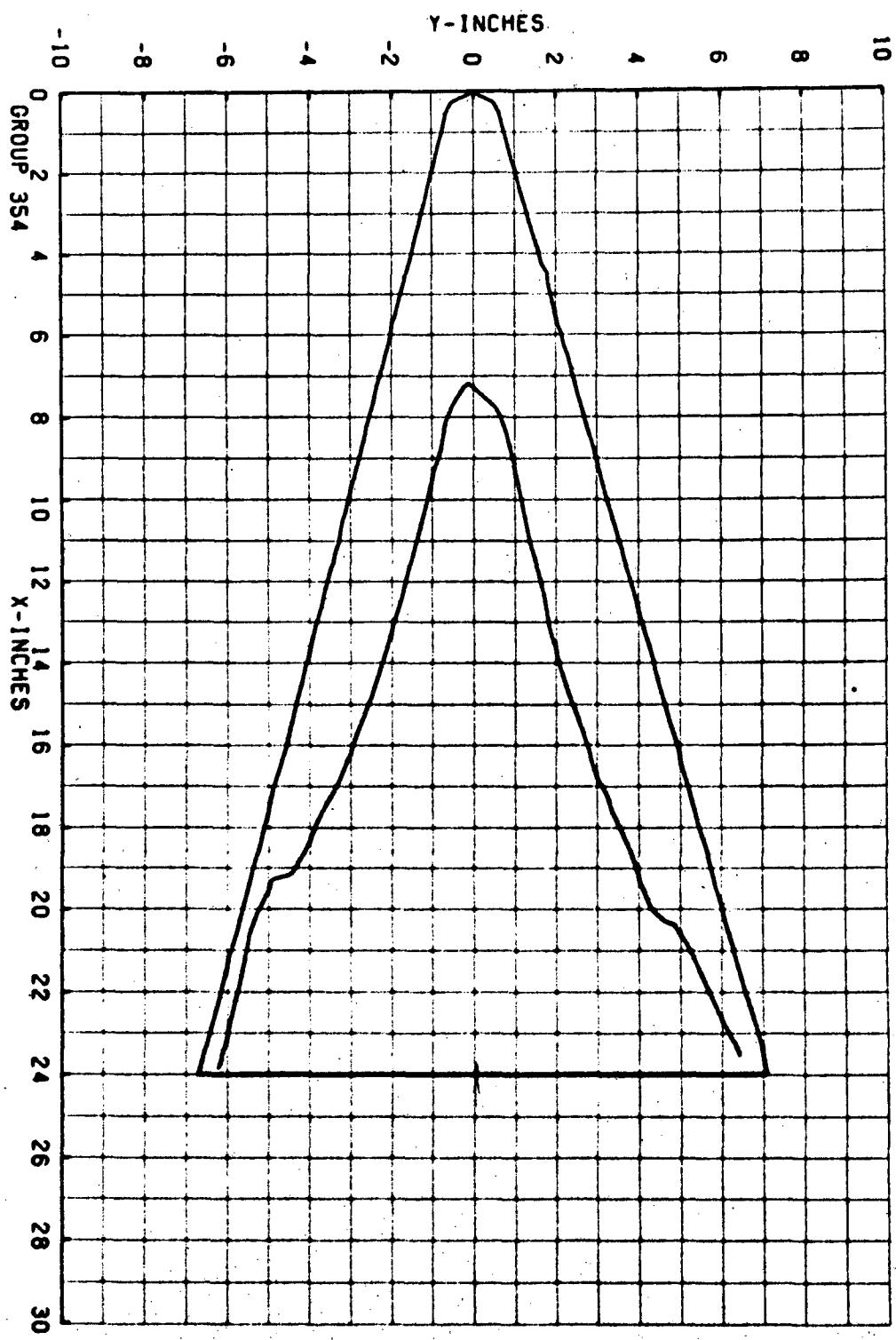
GROUP 129 PIC. NO. 1962 MODEL SURFACE - BOTTOM
MACH 8.00 H/HREF 1.536E-01
ALPHA (DEG) 40.0 HREF 2.208E-02
RE/FT 2.480E 06 CONF LRC-DB

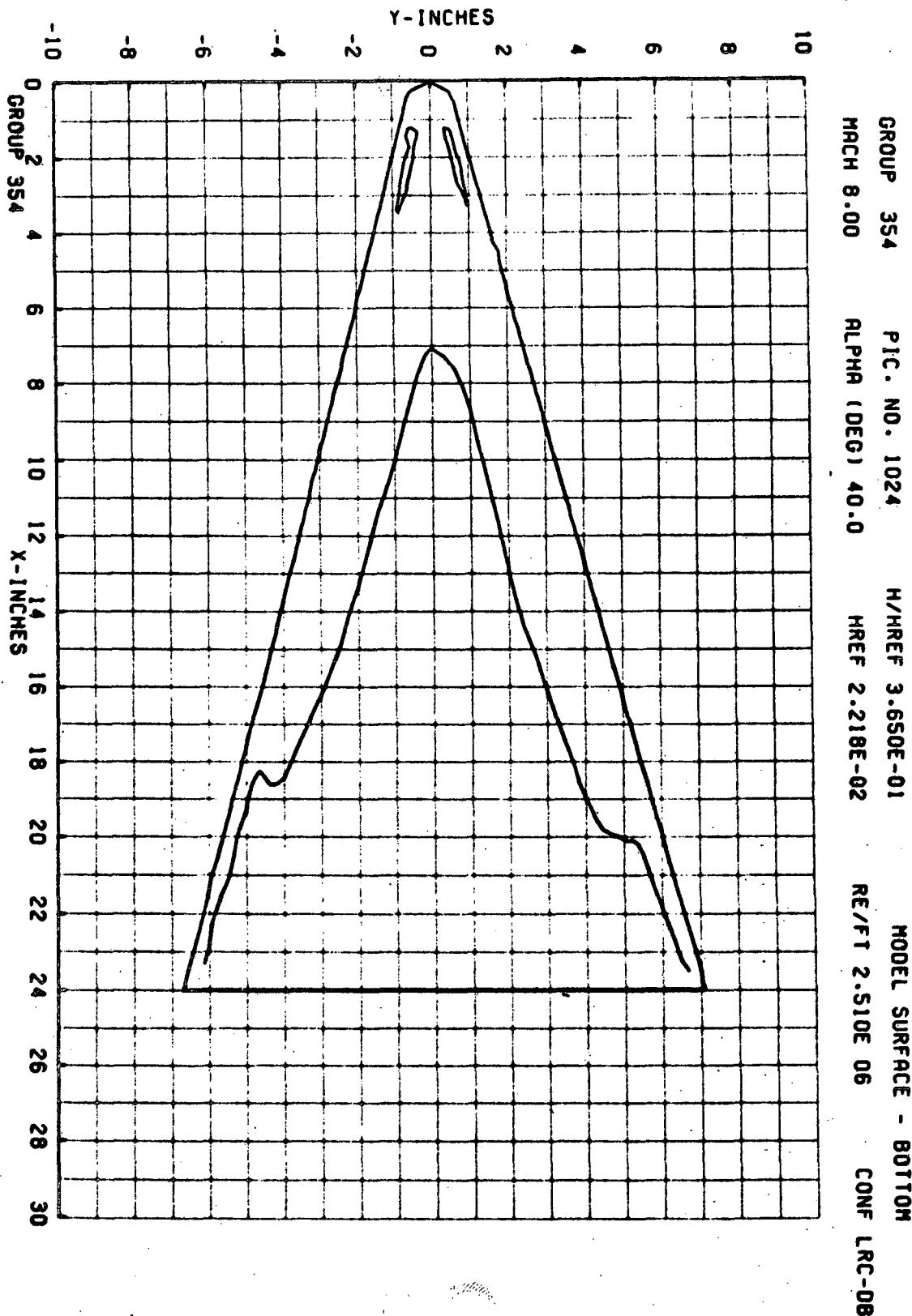


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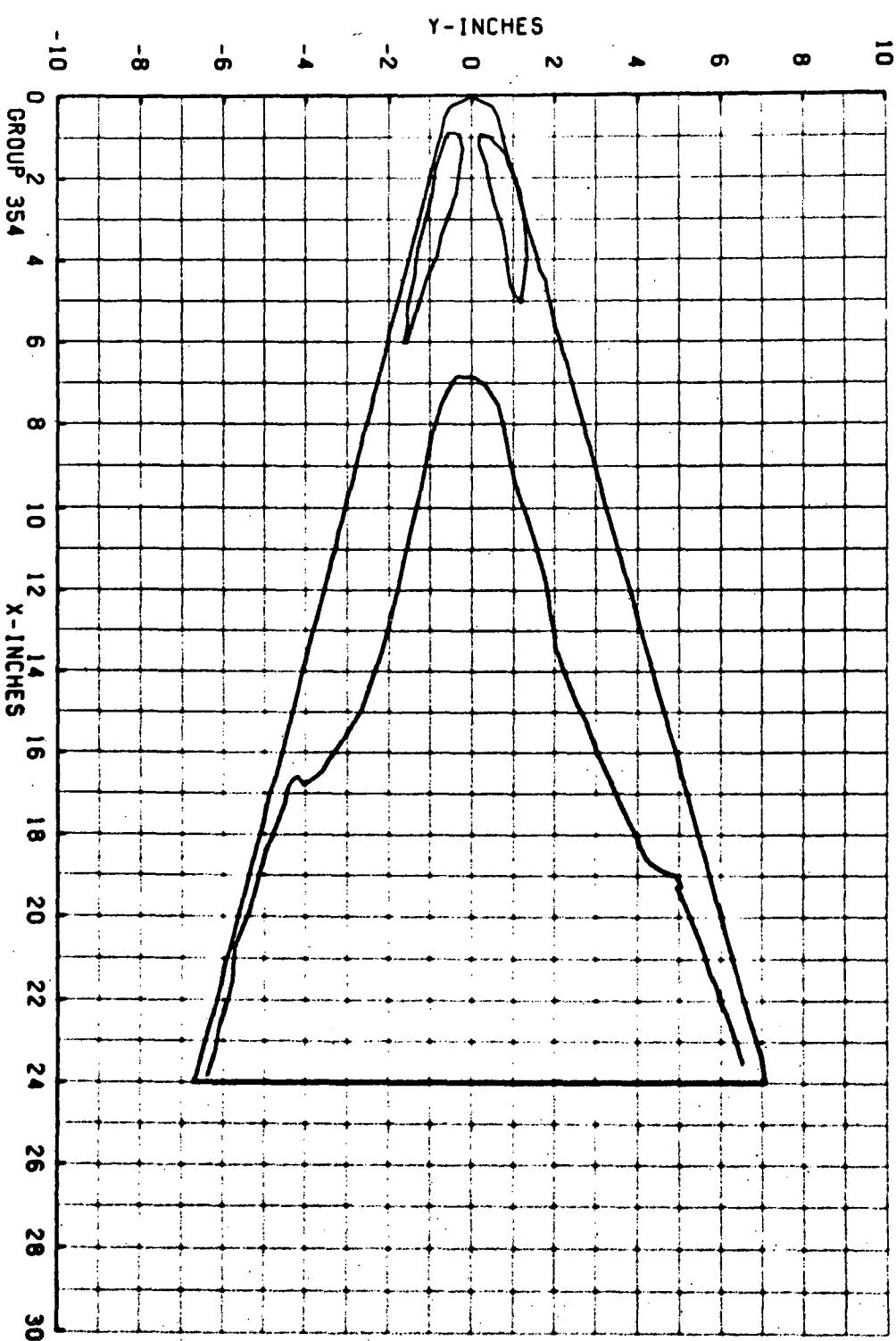


GROUP 354 PIC. NO. 1023 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/HREF 3.963E-01 RE/FT 2.510E-06 CONF LRC-08
ALPHA (DEG) 40.0 H/HREF 2.218E-02

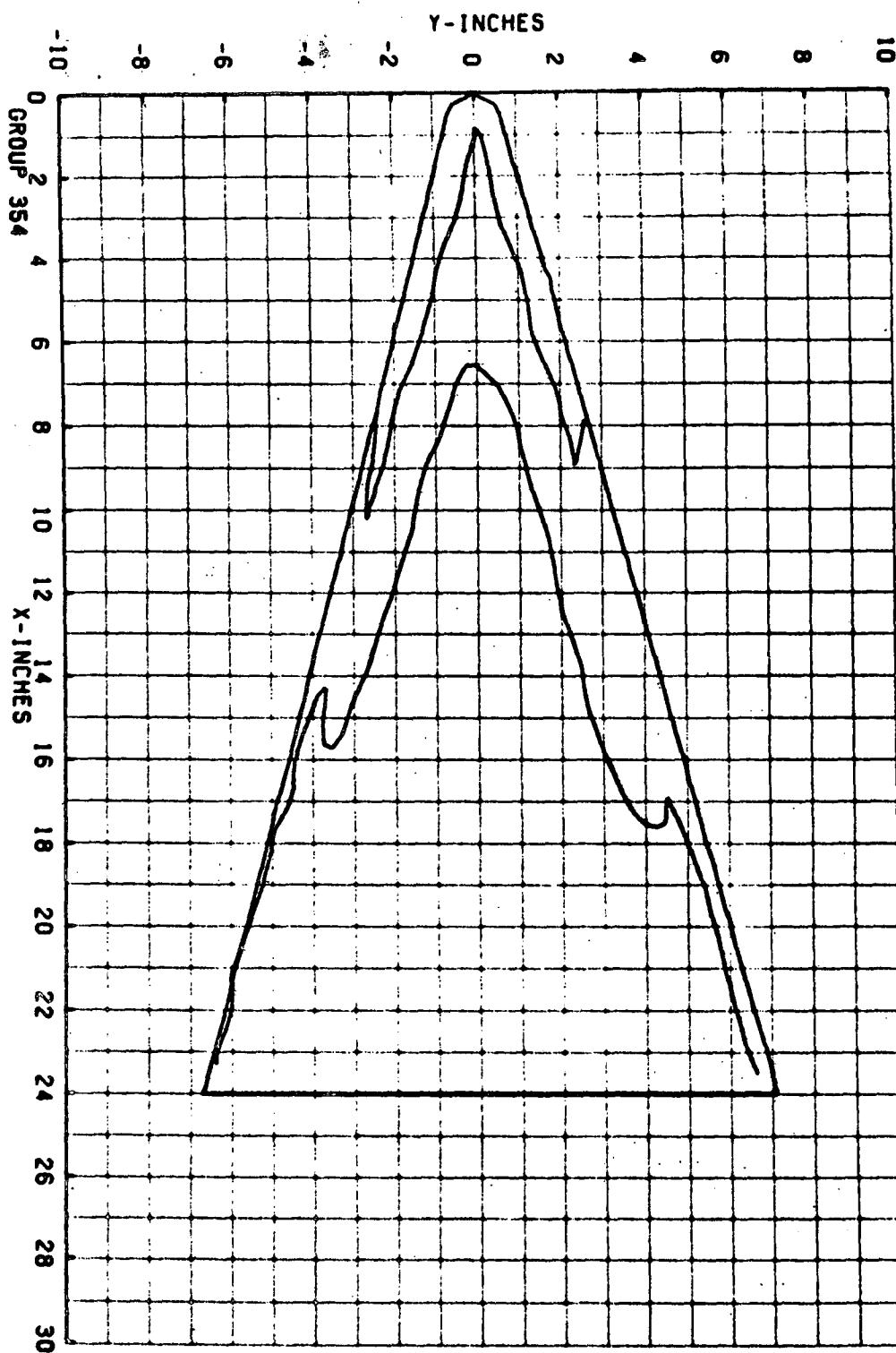




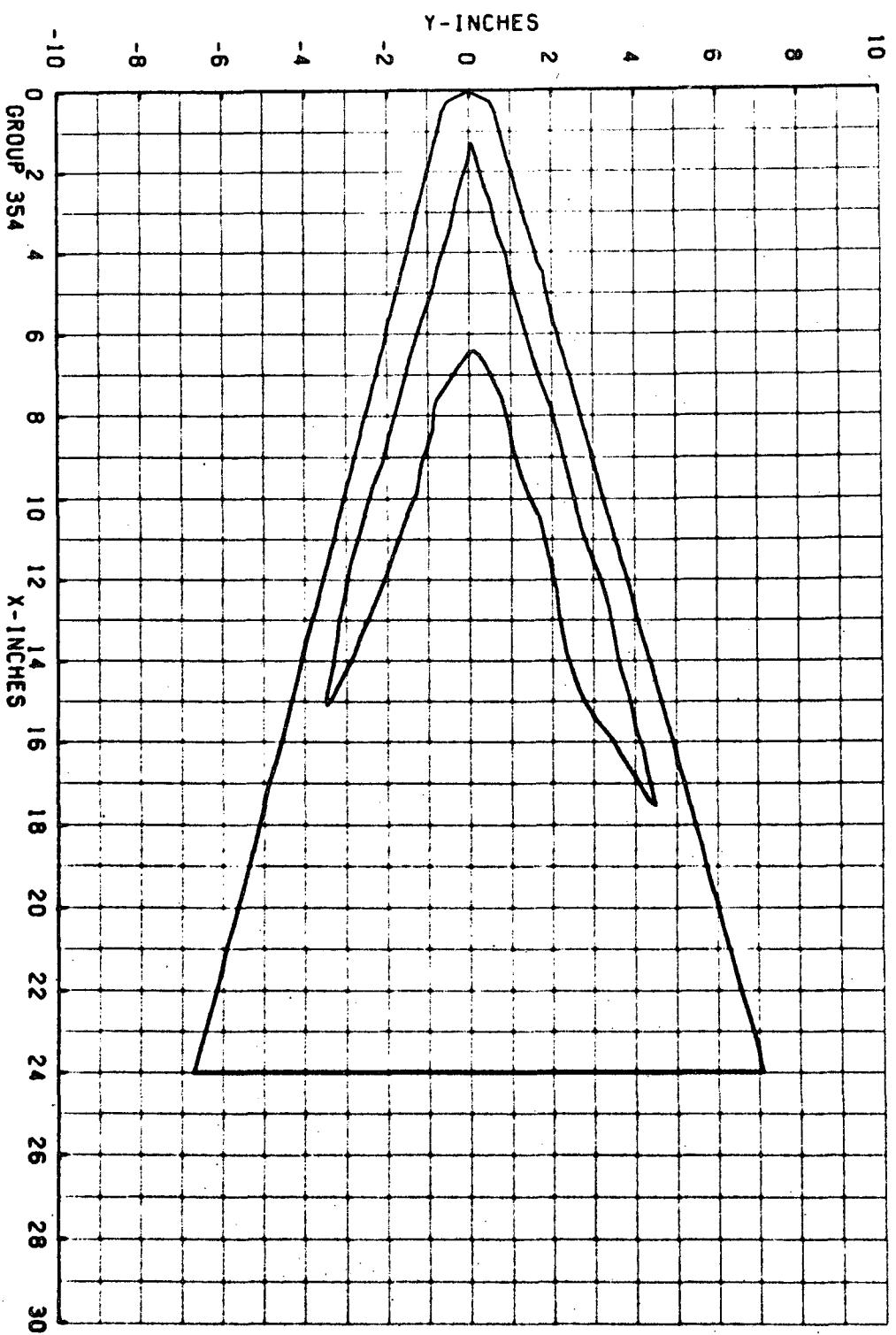
GROUP 354 PIC. NO. 1026 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/HREF 3.215E-01
HREF 2.218E-02 RE/FT 2.510E 06 CONF LRC-08



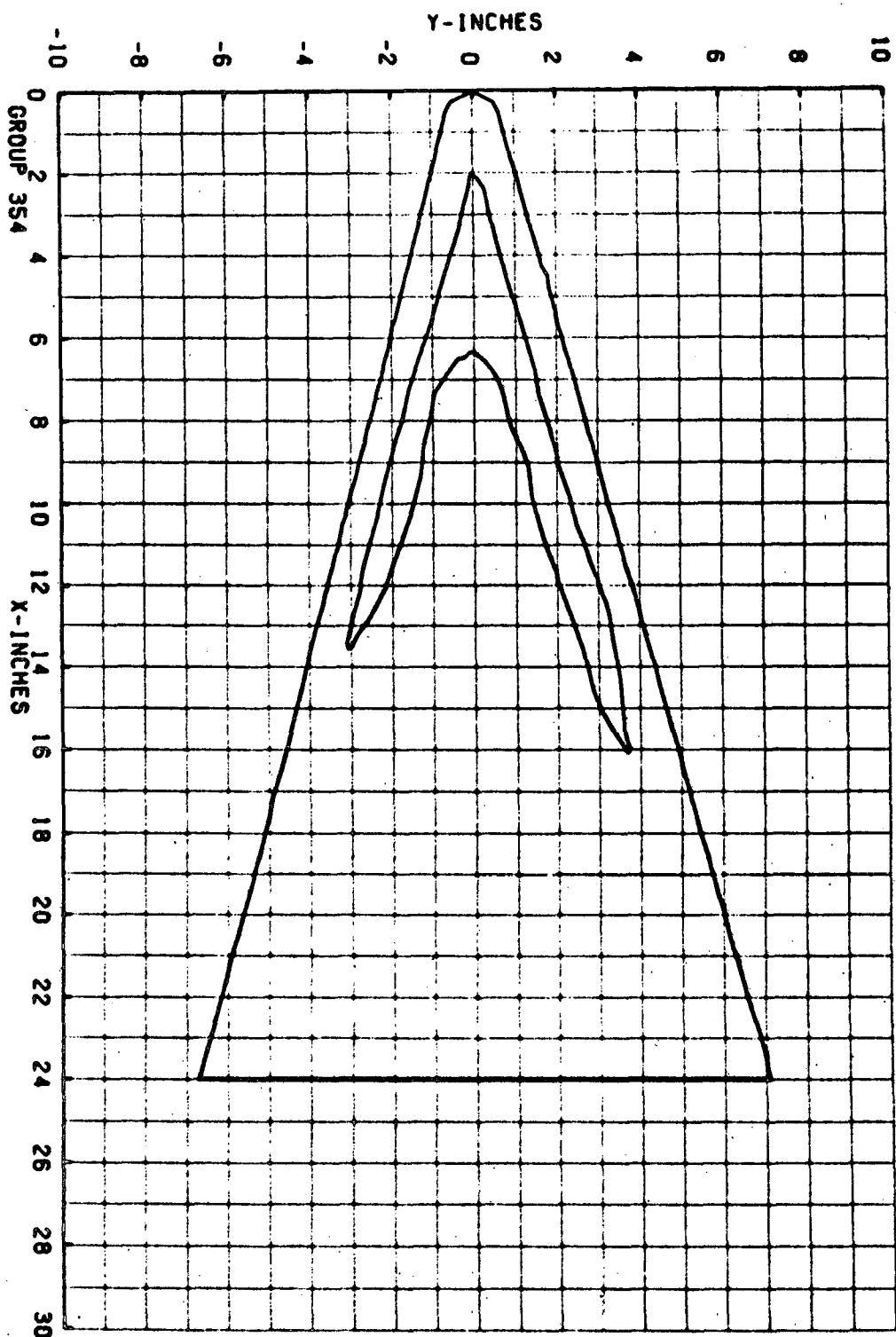
GROUP 354 PIC. NO. 1029 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/HREF 2.787E-01
HREF 2.218E-02 RE/FI 2.510E 06 CONF LRC-08



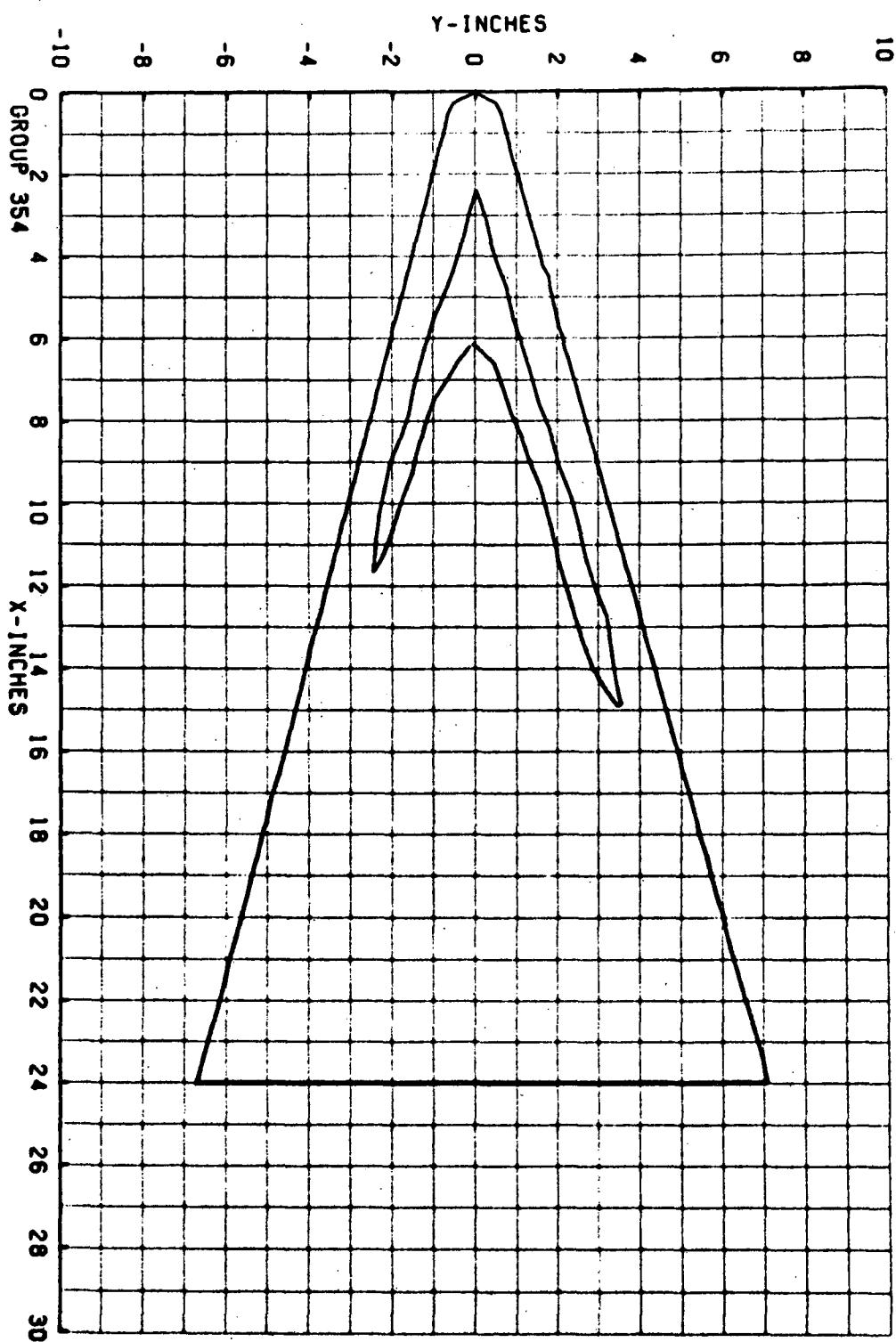
GROUP 354 PIC. NO. 1032 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/REF 2.485E-01
ALPHA (DEG) 40.0 HREF 2.218E-02 RE/FT 2.510E 06 CONF LRC-08



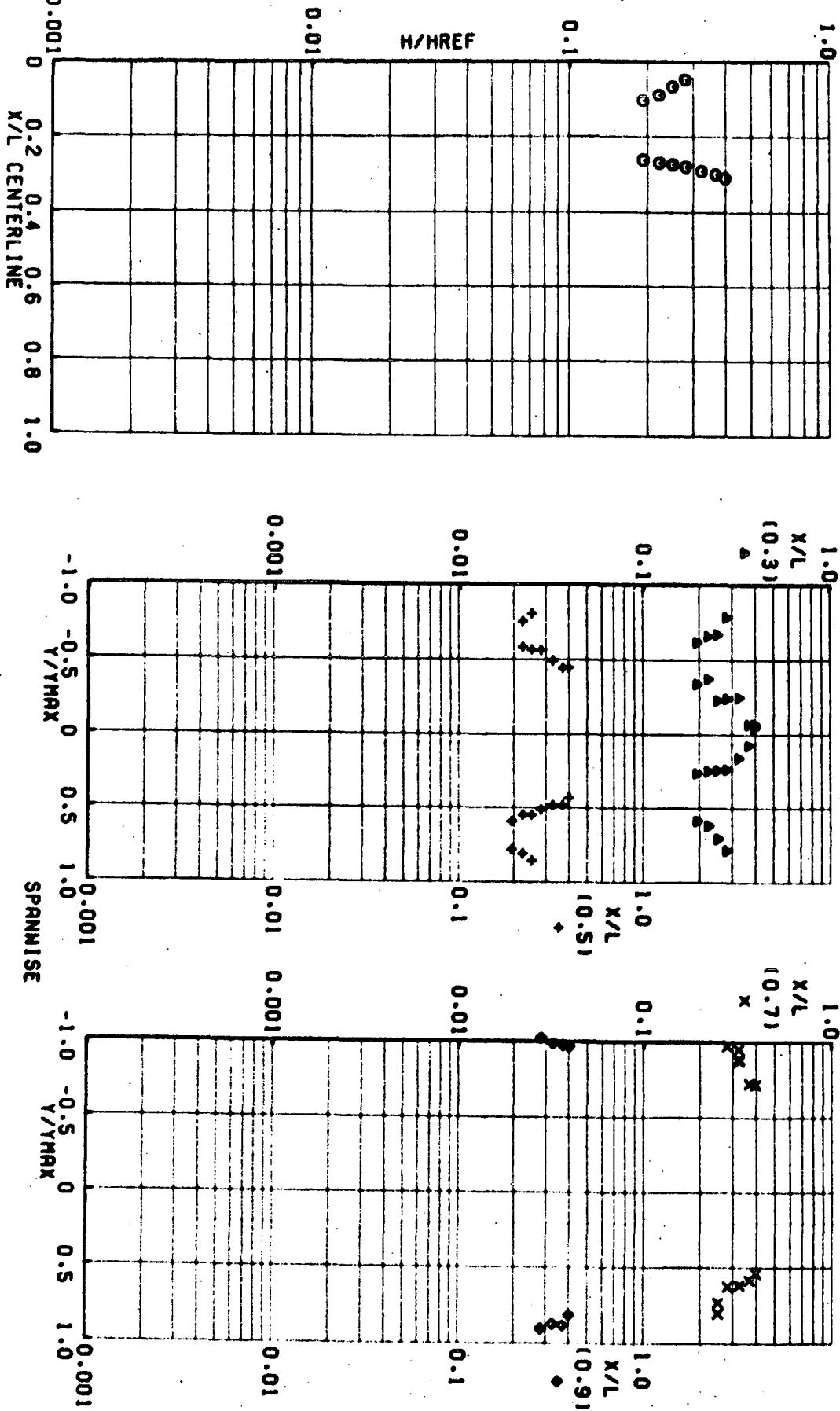
GROUP 354 PIC. NO. 1036 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/HREF 2.207E-01
GROUP 2 354 ALPHAREF 2.218E-02 RE/FT 2.510E 06 CONF LRC-0B



GROUP 354 PIC. NO. 1042 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/HREF 1.923E-01
HREF 2.216E-02 RE/FI 2.510E 06 CONF LRC-DB



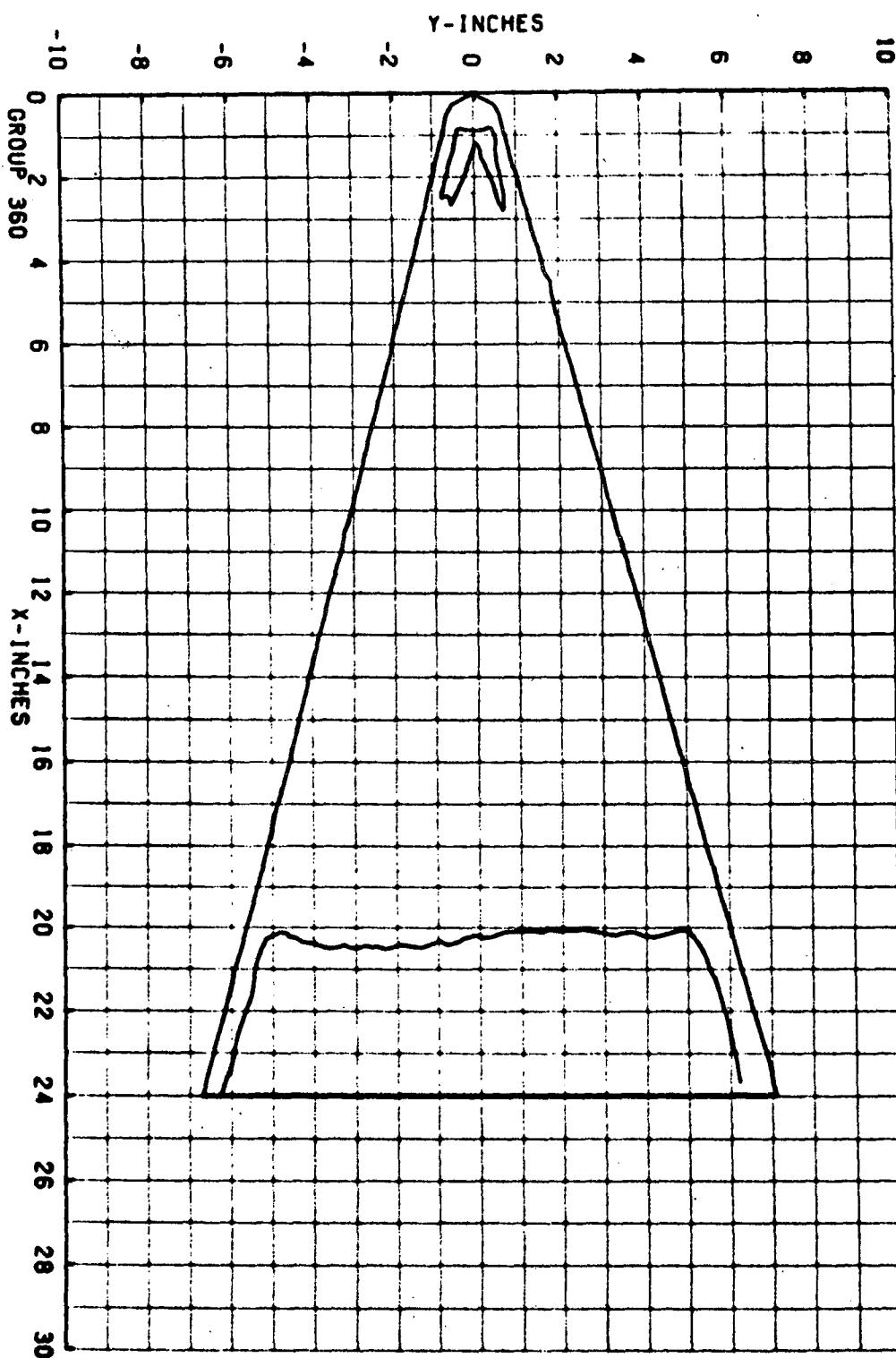
GROUP 354 ALPHA (DEG) 40.0 MREF 2.218E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FT 2.510E 06 CONF LRC-DB



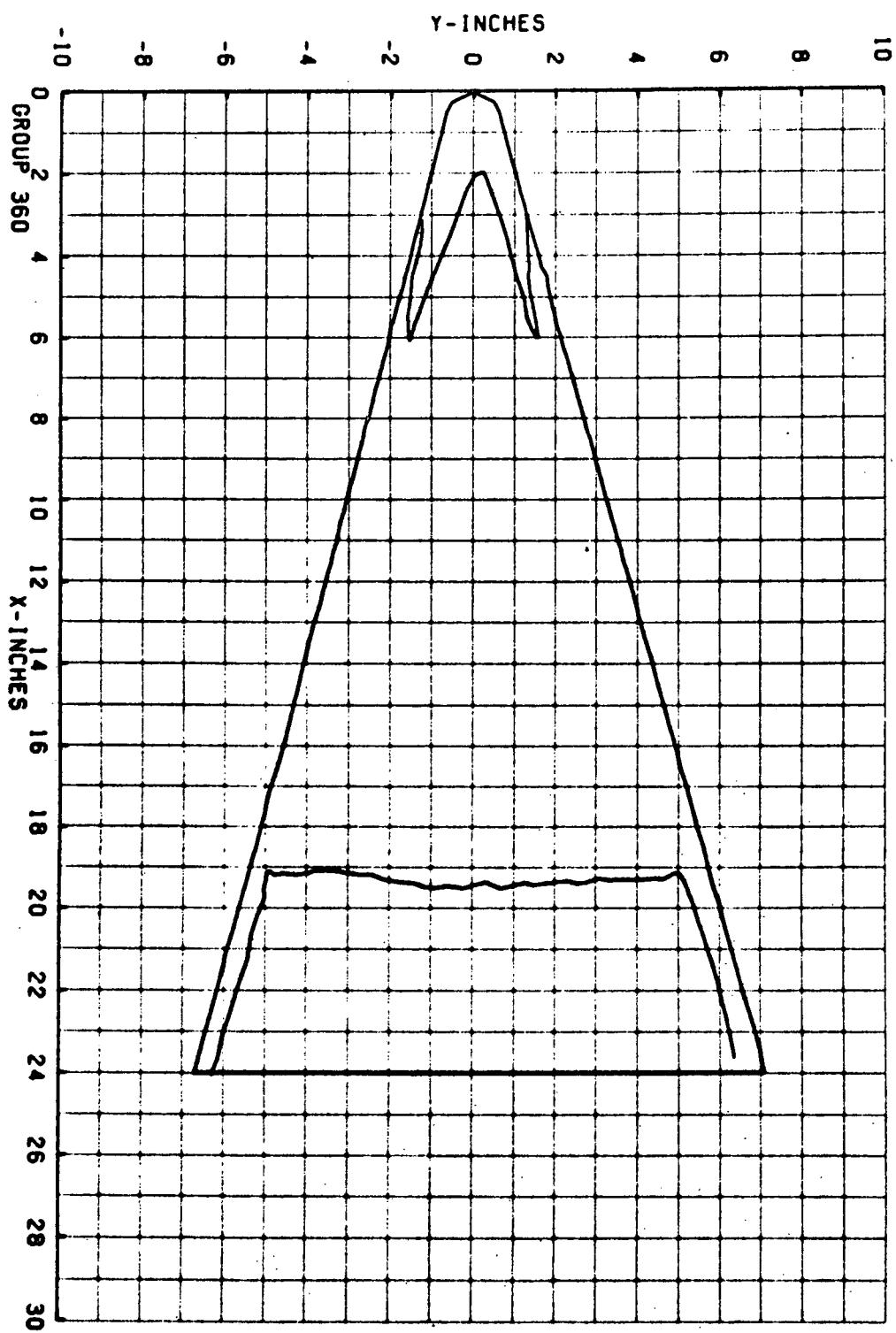
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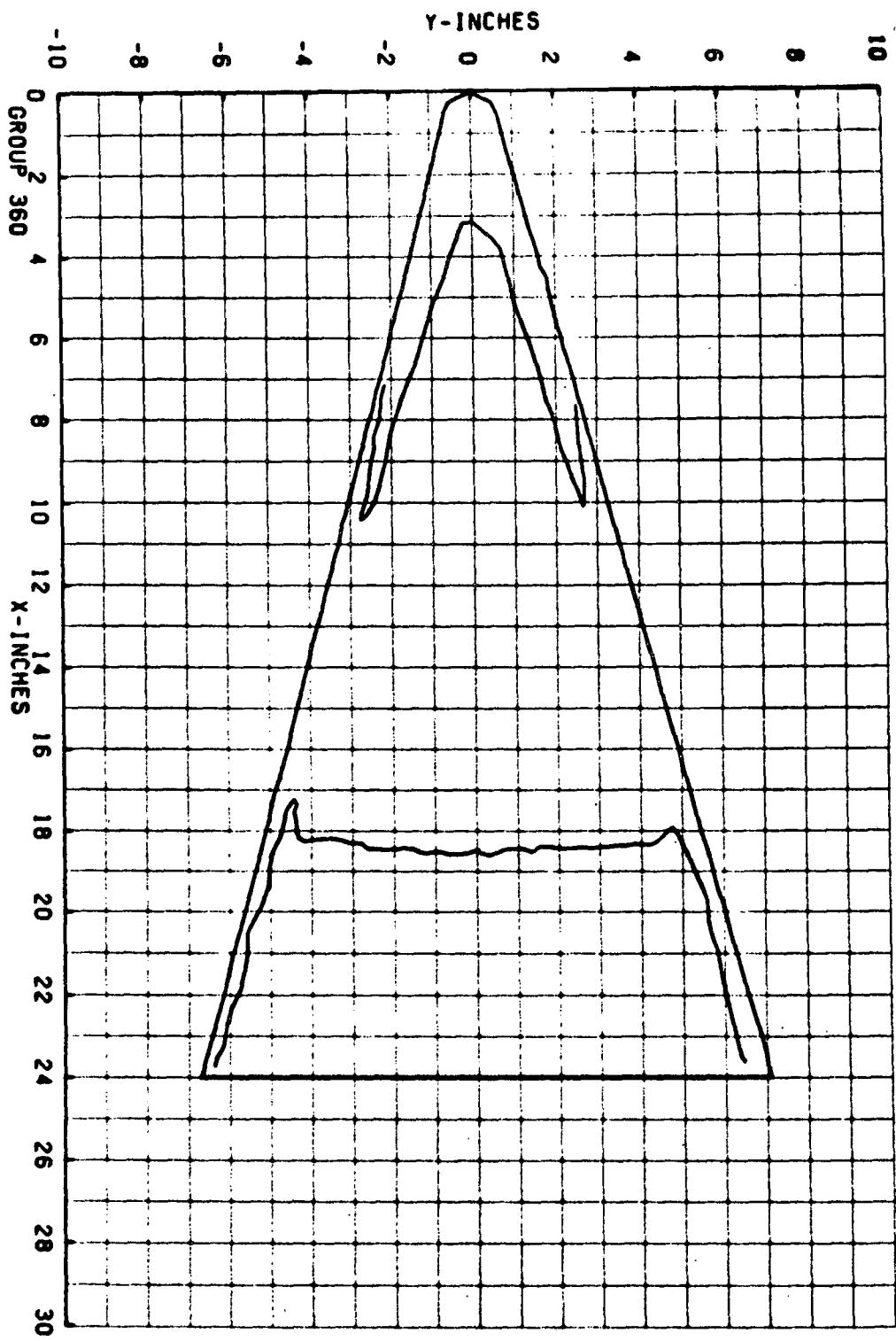
AEDC (AMU INC.) ARNOLD AFS, TENNESSEE
VUN KARMAN GAS DYNAMICS FACILITY
50-INCH HYPERSONIC TUNNEL B

GROUP 360 PIC. NO. 1244 H/HREF 4.462E-01
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.208E-02
RE/FT 2.530E 06 CONF LRC-DB



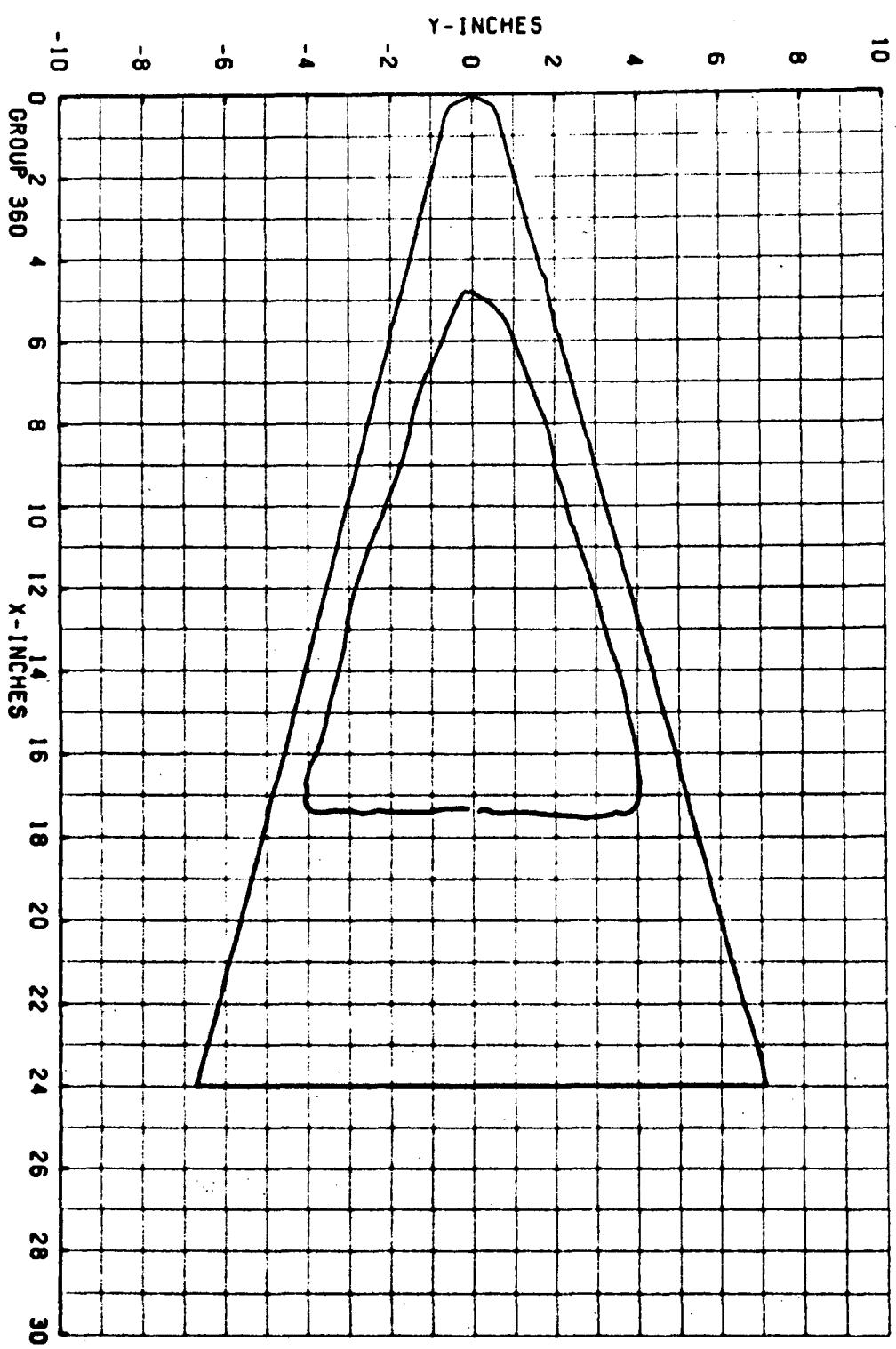
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MACH 8.00 ALPHA (DEG) 60.0 HREF 2.208E-02
MODEL SURFACE - BOTTOM
RE/FT 2.530E 06 CONF LRC-DB



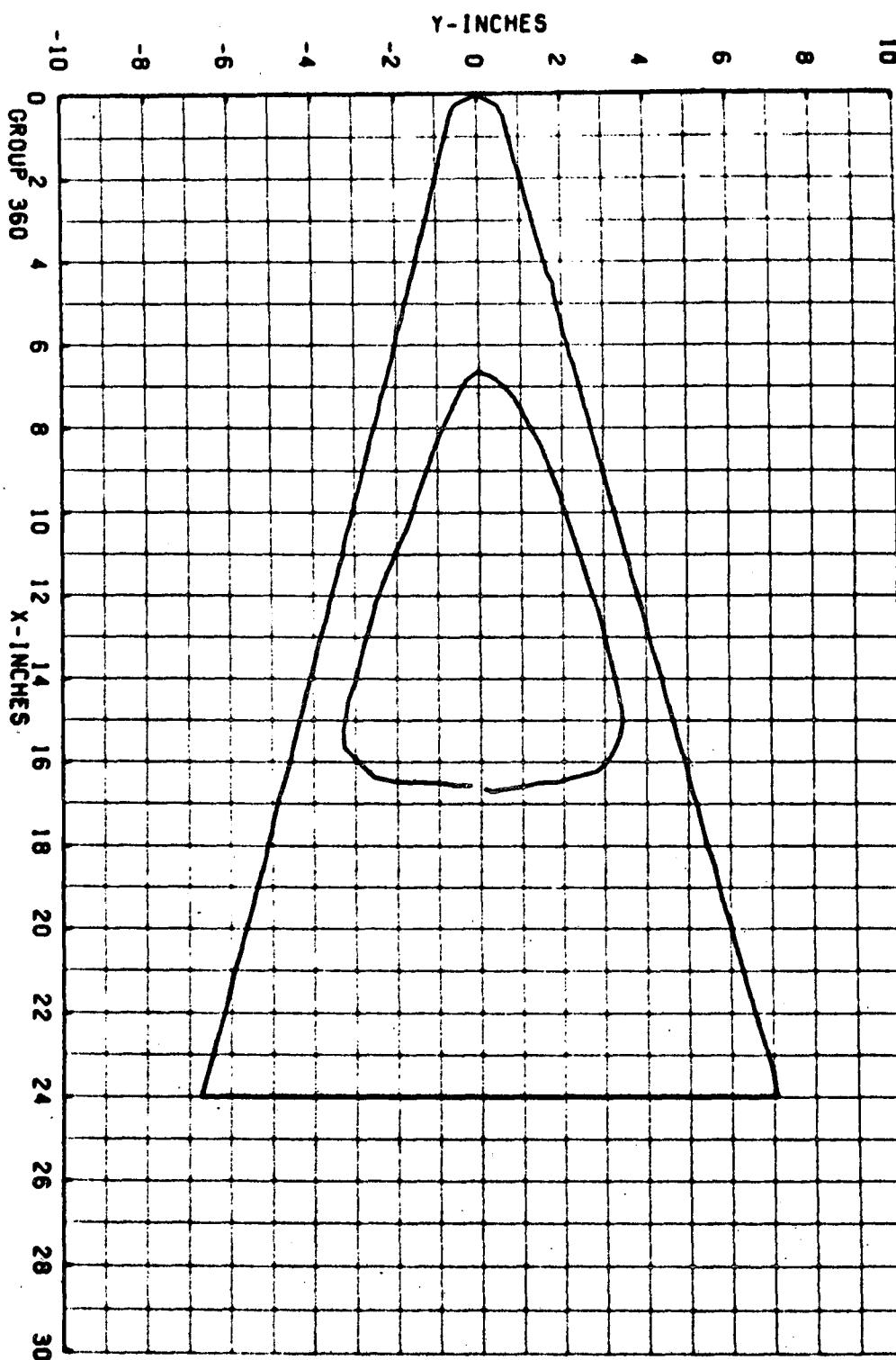


GROUP 360 PIC. NO. 1250 H/HREF 2.967E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.208E-02 RE/FT 2.530E 06 CONF LRC-08

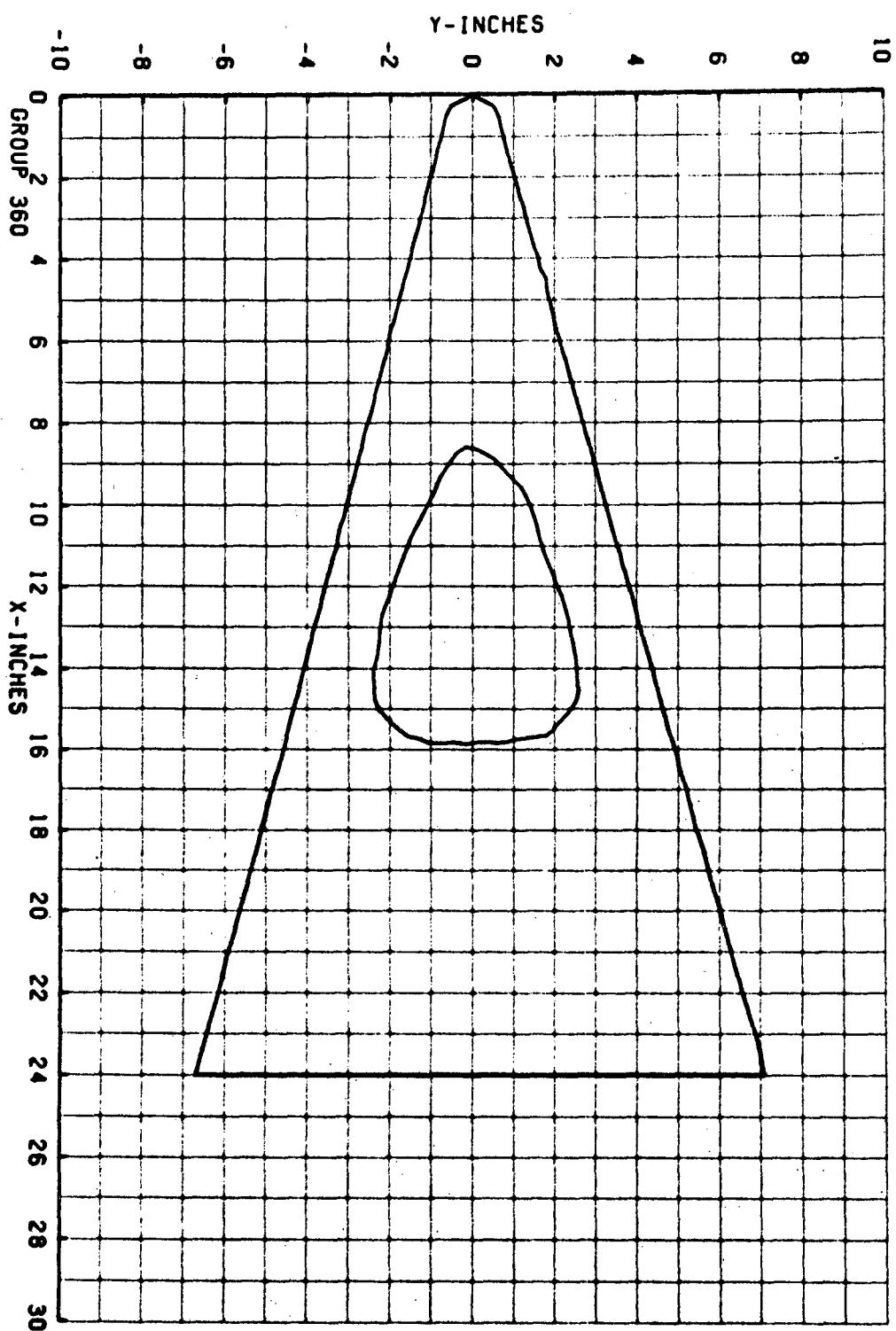
GROUP 360 PIC. NO. 1254 H/HREF 2.533E-01
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.208E-02
MODEL SURFACE - BOTTOM
RE/FT 2.530E 06 CONF LRC-08



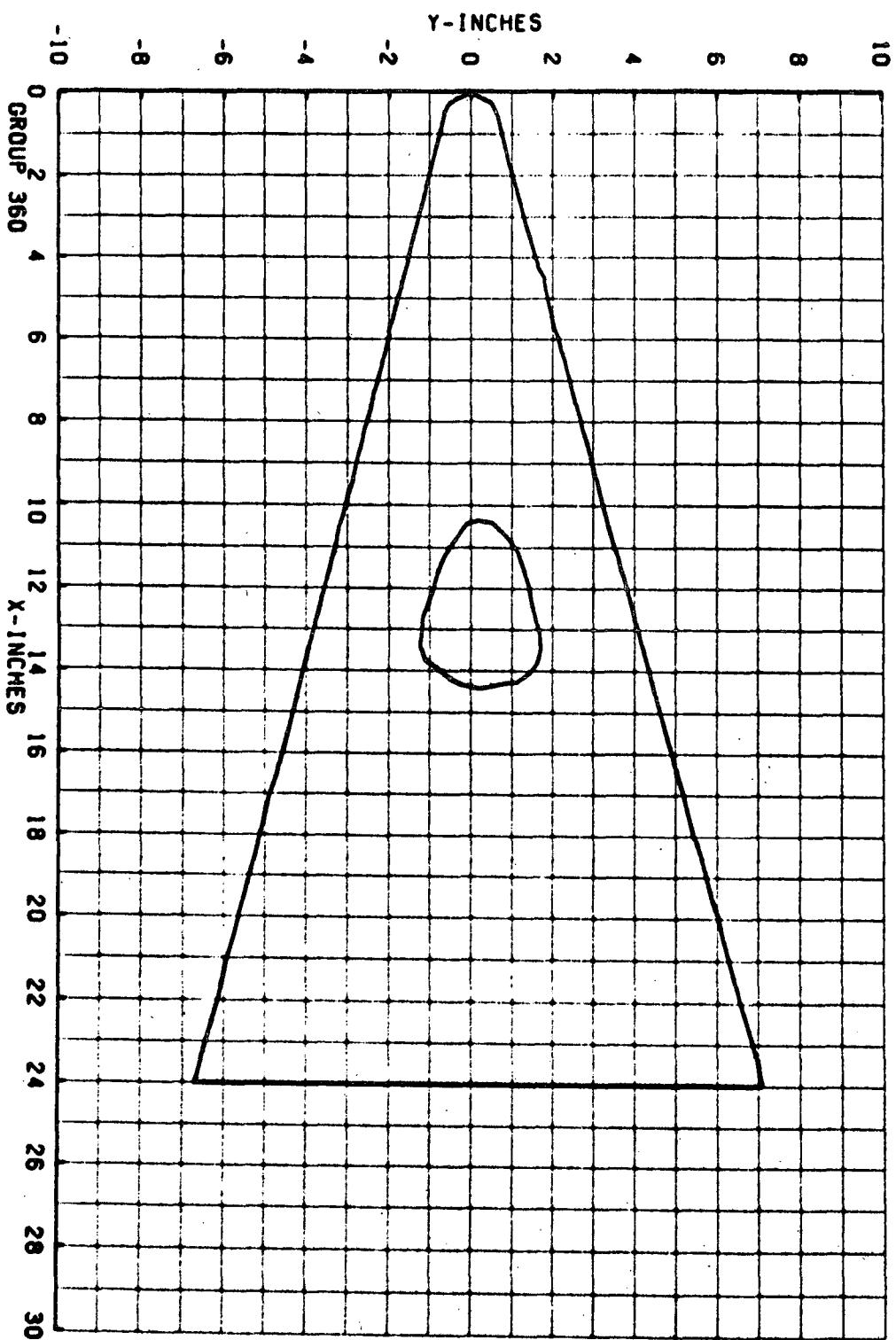
GROUP 360 PIC. NO. 1260 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 60.0 H/HREF 2.136E-01
HREF 2.208E-02 RE/FT 2.530E 06 CONF LRC-08



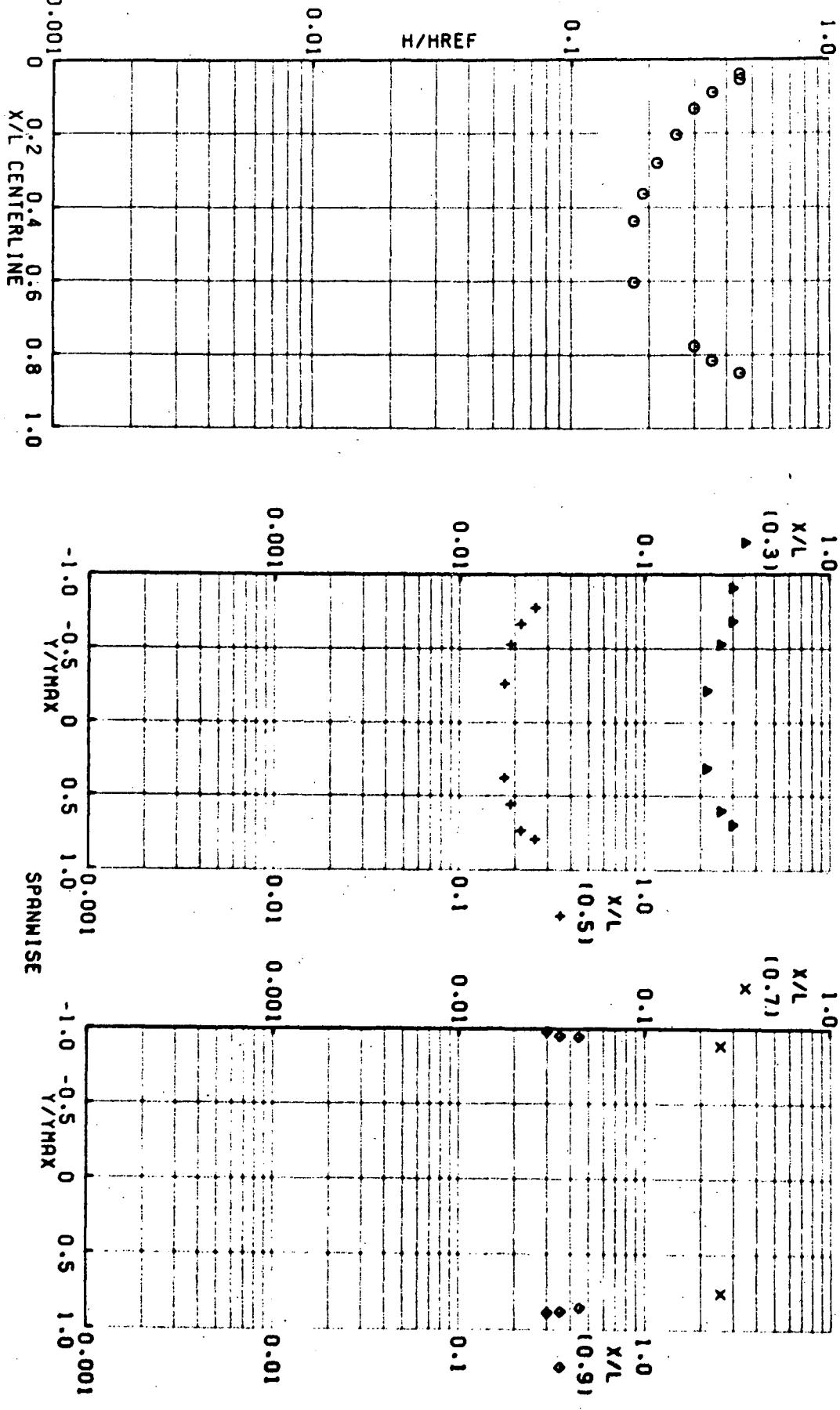
GROUP 360 PIC. NO. 1266 H/HREF 1.881E-01
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.208E-02
GROUP 2 360 RE/FT 2.530E 06 CONF LRC-08
MODEL SURFACE - BOTTOM



GROUP 360 PIC. NO. 1269 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 60.0 H/HREF 1.741E-01
ALPHA (DEG) 60.0 HREF 2.208E-02 RE/FT 2.530E 06
CONF LRC-DB



GROUP 360 ALPHA (DEG) 60.0 HREF 2.208E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FT 2.530E 06 CONF LRC-08



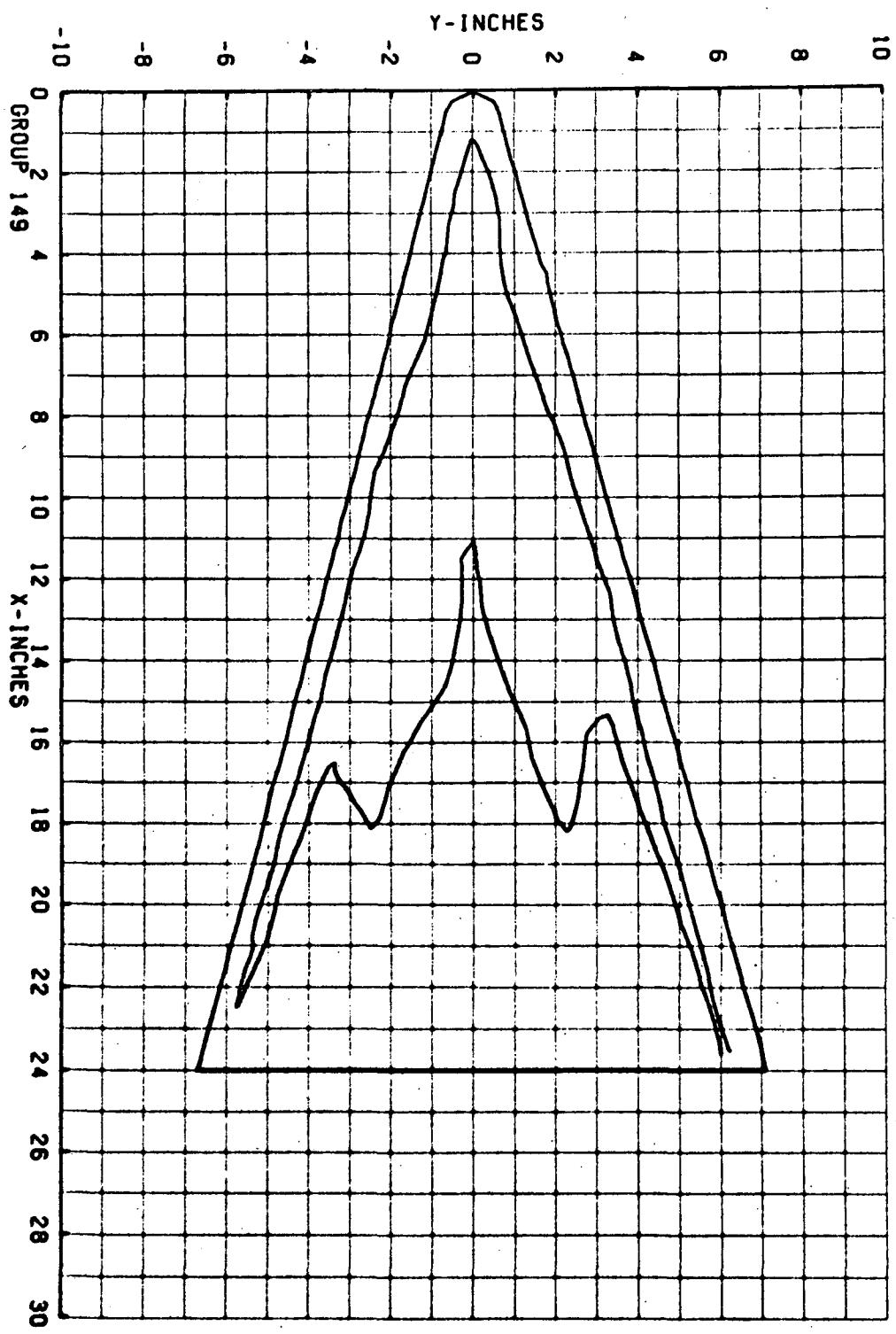
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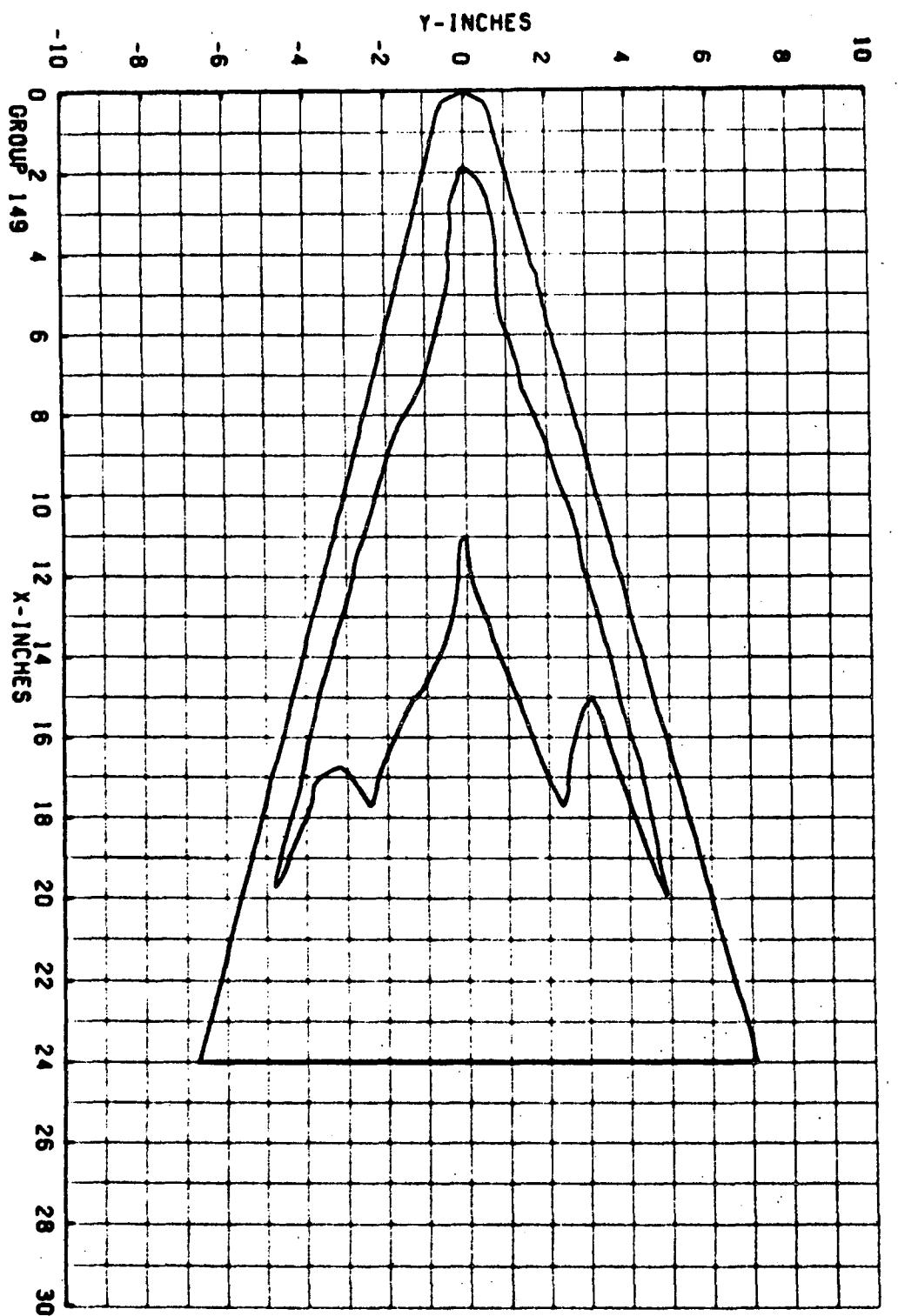
AEDC(LARO INC.) ARNOLD AFS • TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50-INCH HYPERSONIC TUNNEL B.
VT1162

ALPHA-MODEL ALPHA-MODEL ALPHA-SECTOR ALPHA-PREBEND ROLL-MODEL YAW
149 11 LAC-DB 8.00 859.3 1320 19.97 3.03 -23.00 180.00 0
1-INF P-INF 0-INF V-INF RHO-INF MU-INF REF/HREF STREF
(DEG R) (PSIA) (PSIA) (FT/SEC) (SLUGS/FT³) (LB-SEC/12) (FT-1) (R= .056FT) (H= .056FT)
95.7 .088 3.943 3834 7.218E-05 2.703E-08 3.08E-06 2.152E-02 1.159E-02
CAVERA PAINT TEMP (DEG F) INITIAL TEMP (DEG F) SQUARE ROOT (RHOXCXK)
TOP(1) 200 200 AVERAGE IM = 95 =0.008(SQUARE ROOT DEL TIME) * 0.11
SIDE(S)
BOT(B)

PIC NC TIME DELAY H(110) H(110)/HREF H(.910) H(.910)/HREF H(.8510) H(.8510)/HREF ST(110) MODEL TEMP F
1 2602 (200) 6.80 5.76 5.15E-03 .1869 6.386E-13 .2319 7.263E-03 .2637 2.165E-03 0 0 95 0
1 2606 (200) 8.90 7.86 4.25E-03 .1543 5.273E-03 .1915 5.597E-03 .2178 1.789E-03 0 0 96 0

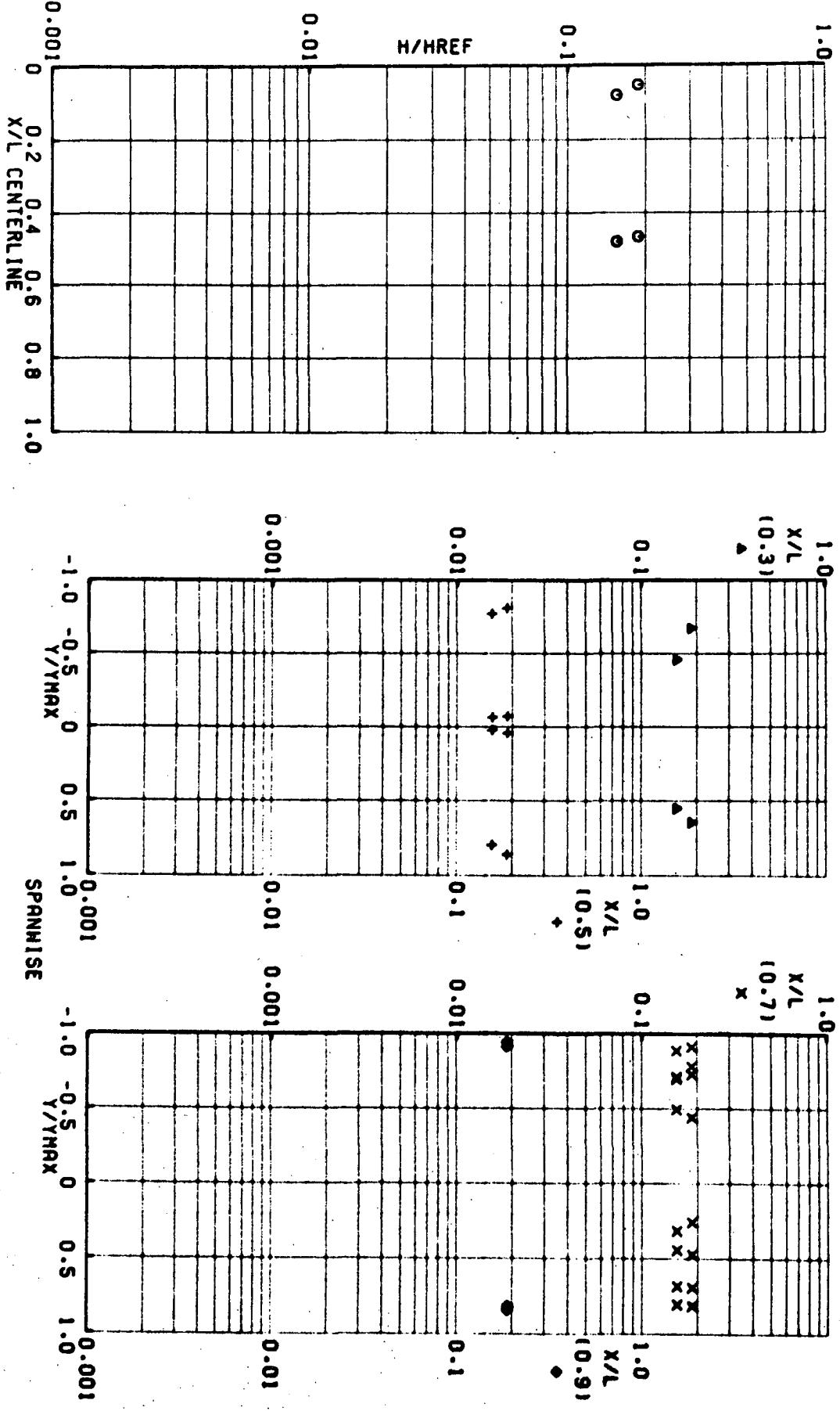
GROUP 149 PIC. NO. 2602 H/HREF 1.869E-01
MACH 8.00 ALPHA (DEG) 20.0 HREF 2.757E-02
RE/FT 3.840E 06 CONF LRC-08
MODEL SURFACE - BOTTOM





GROUP 149 PIC. NO. 2606 H/HREF 1.543E-01
MACH 8.00 ALPHA (DEG) 20.0 HREF 2.757E-02
RE/FI 3.840E 06 CONF LRC-DB
MODEL SURFACE - BOTTOM

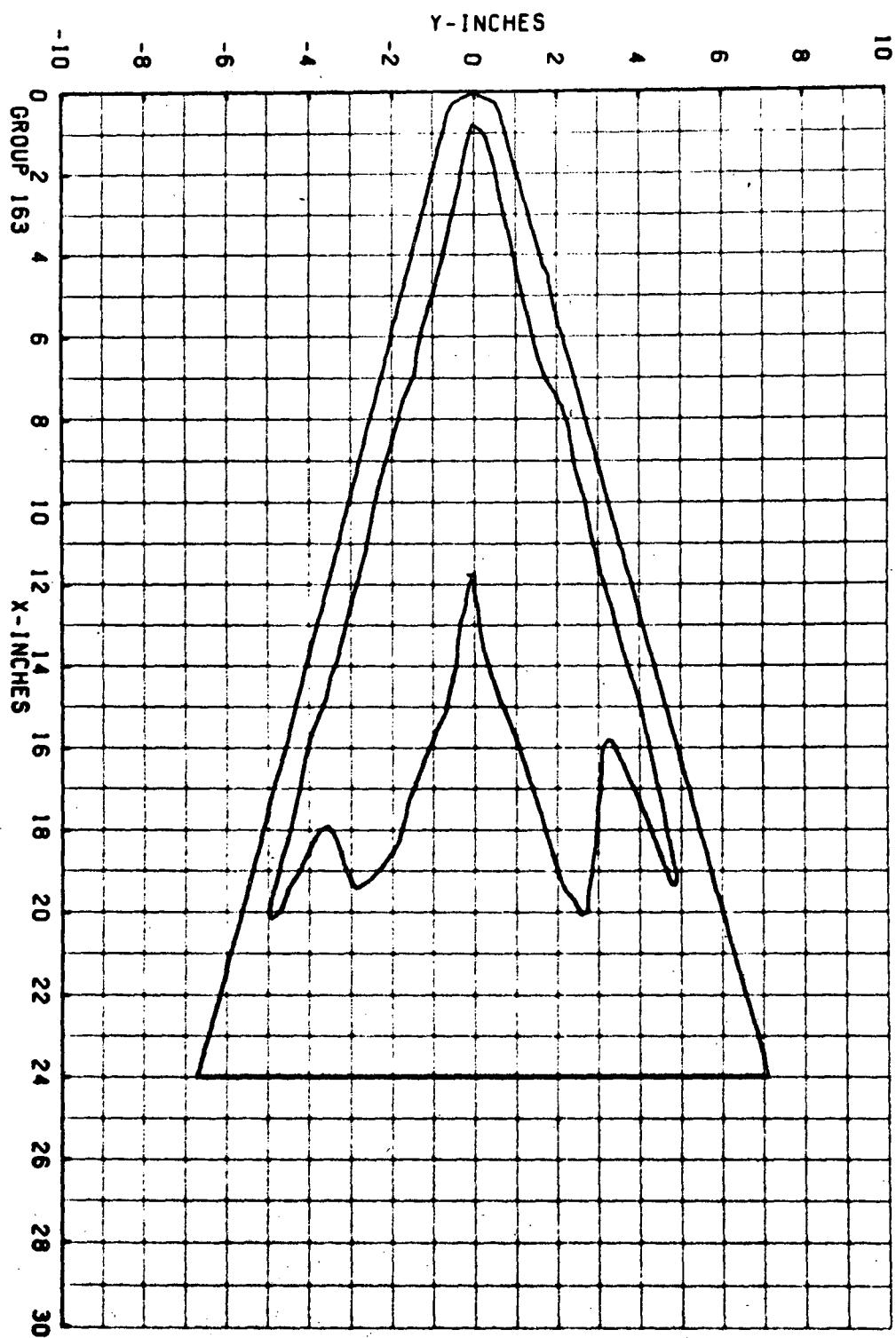
GROUP 149 ALPHA (DEG) 20.0 MREF 2.757E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FT 3.840E 06 CONF LRC-0B

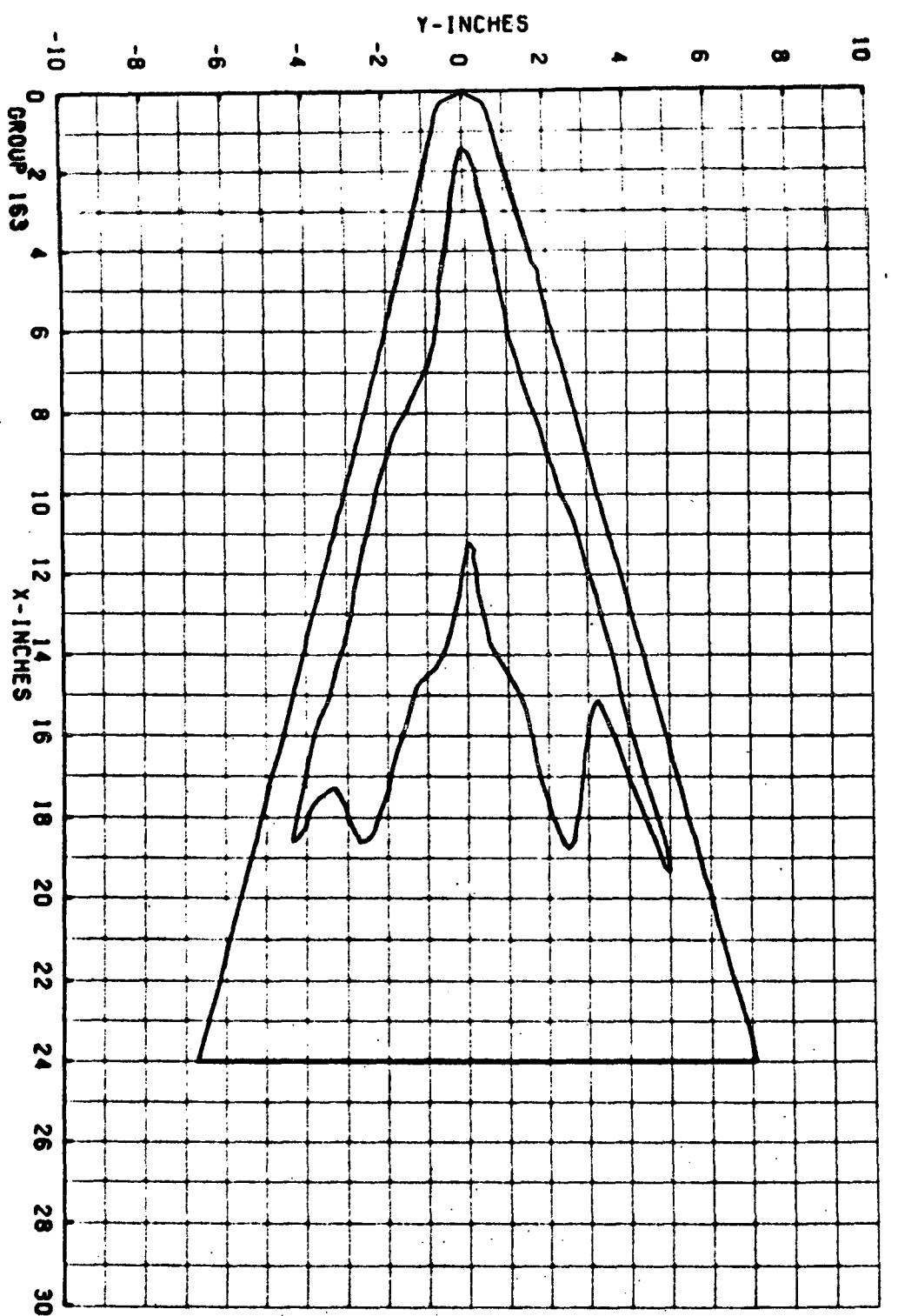


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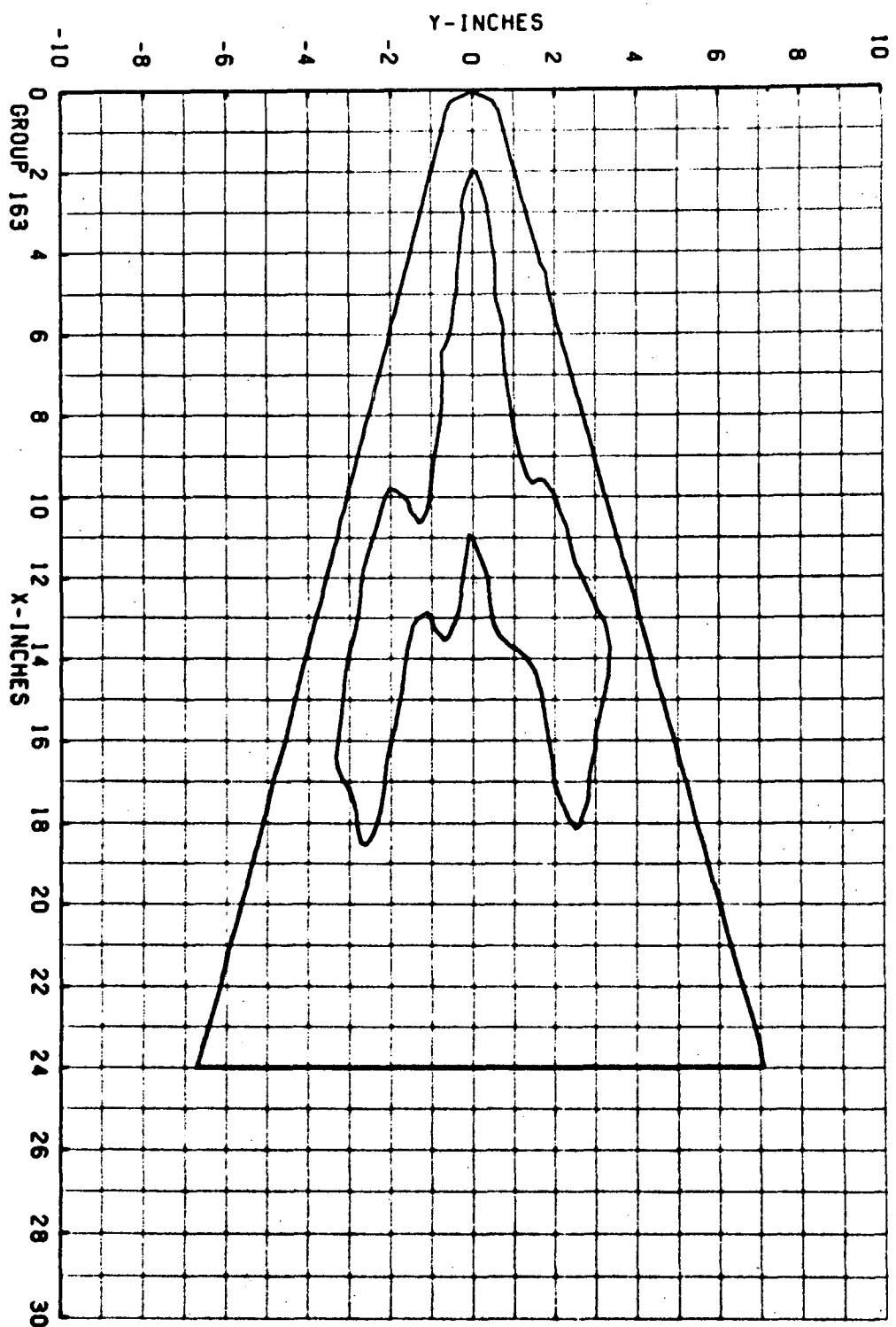
AEDC (IARD, INC.) ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL B
VT1162

GROUP 163 PIC. NO. 3298 M/HREF 2.026E-01
MACH 8.00 ALPHA (DEG) 20.0 HREF 2.769E-02
MODEL SURFACE - BOTTOM
HREF 3.660E 06 CONF LRC-08

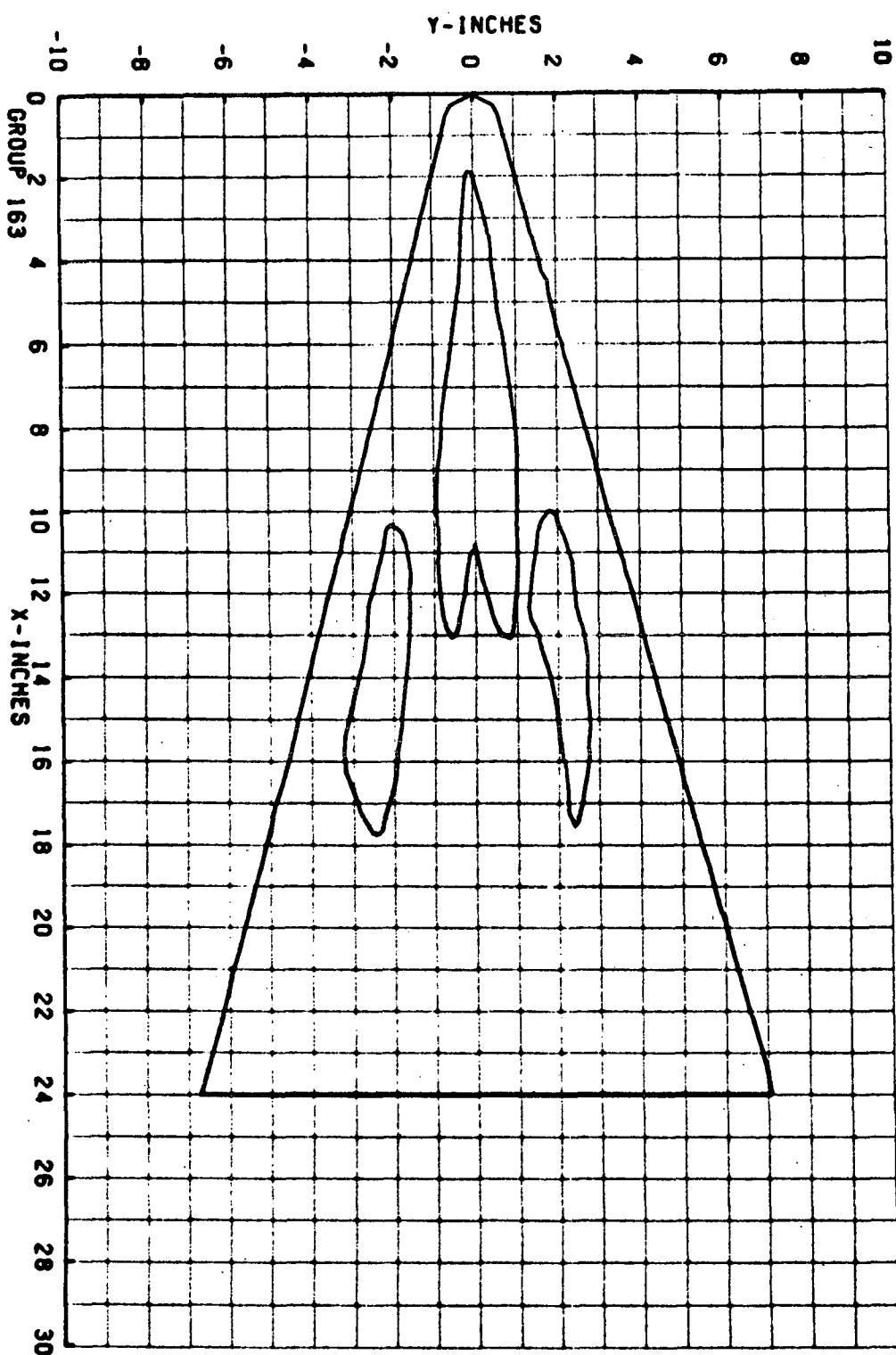




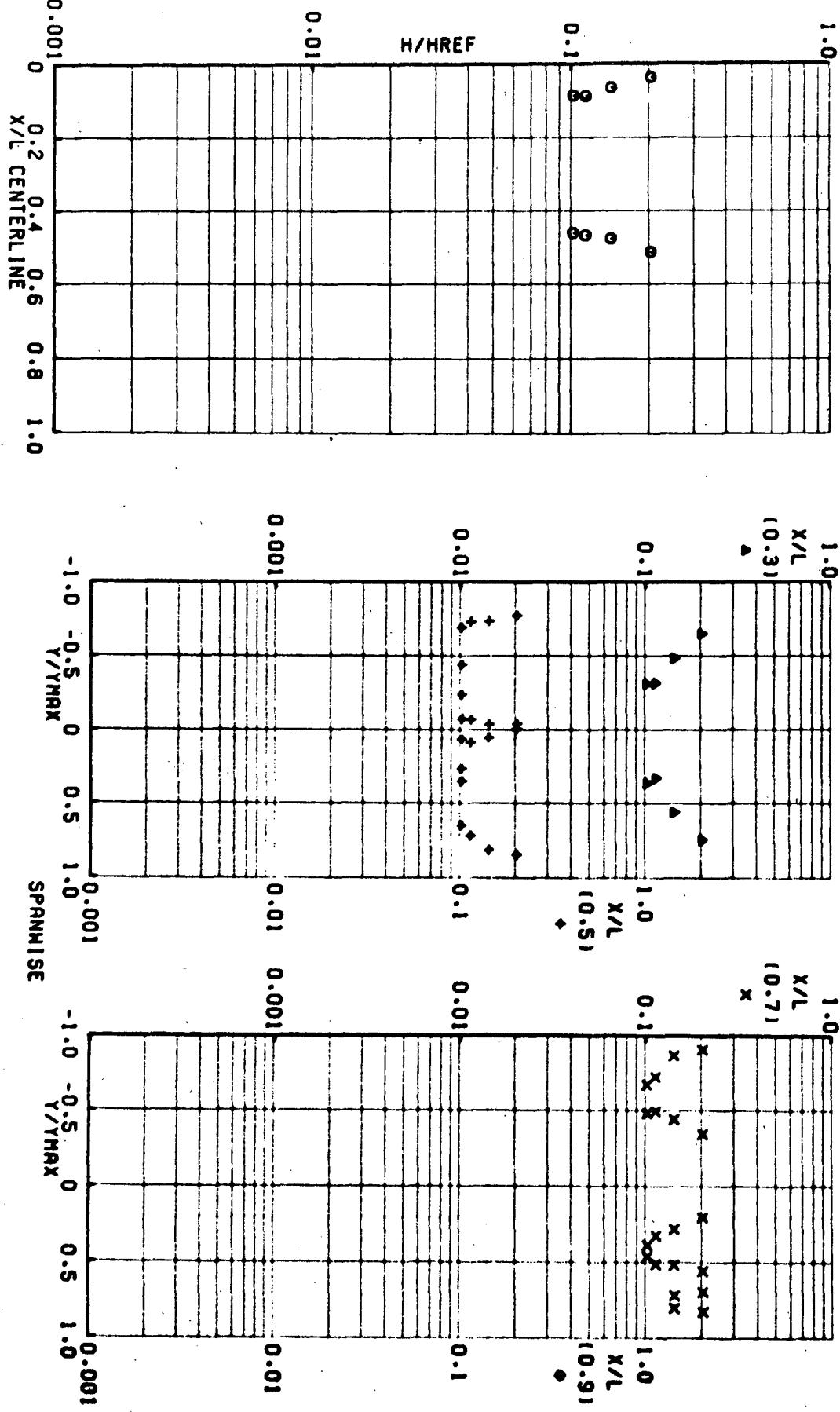
GROUP 163 PIC. NO. 3308 H/HREF 1.134E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 20.0 HREF 2.769E-02 RE/FT 3.660E 06 CONF LRC-08



GROUP 163 PIC. NO. 3311 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 20.0 H/HREF 1.017E-01
HREF 2.769E-02 RE/FT 3.660E 06 CONF LRC-08

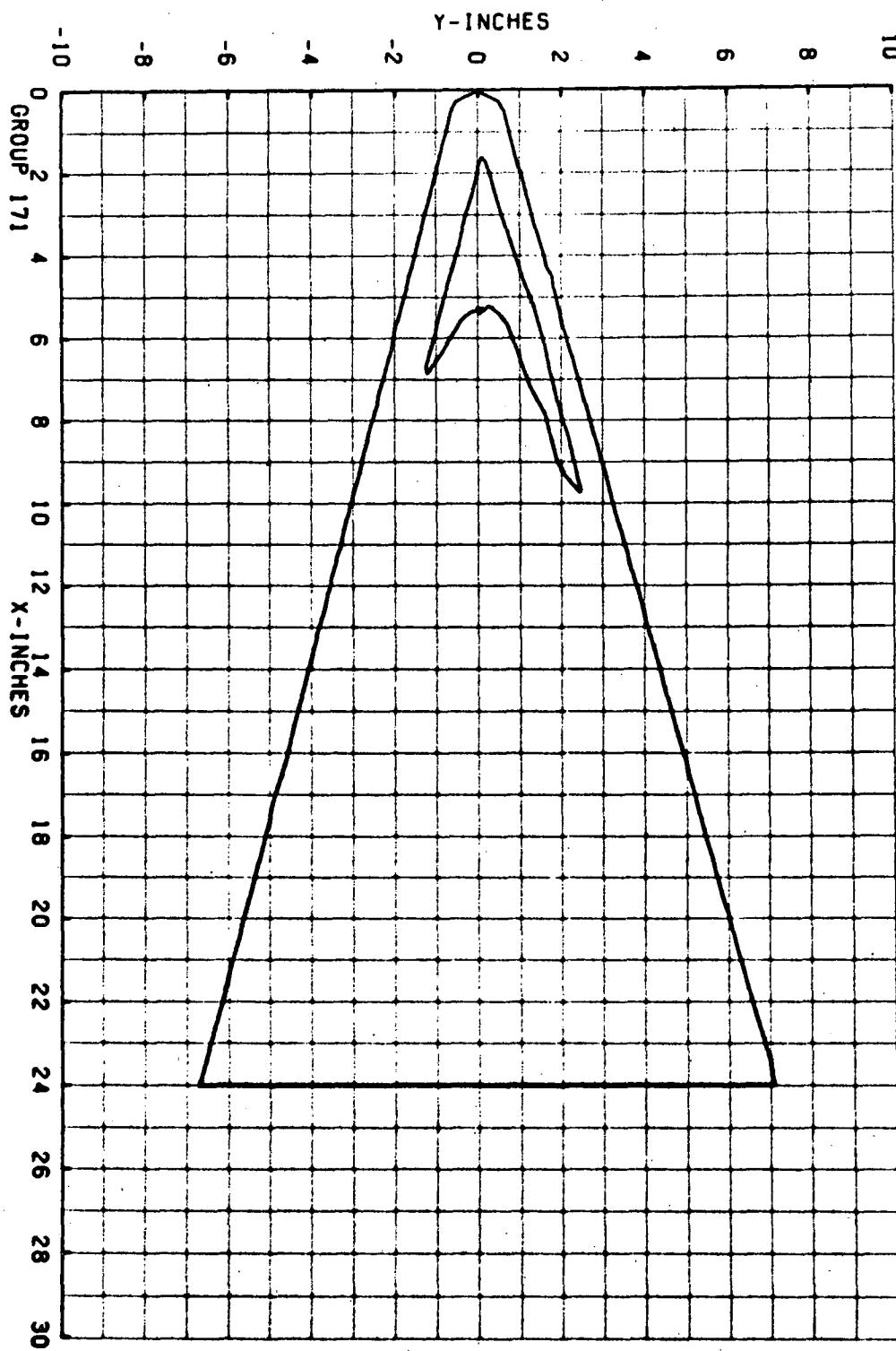


GROUP 163 ALPHA (DEG) 20.0 HREF 2.769E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FT 3.660E 06 CONF LRC-DB



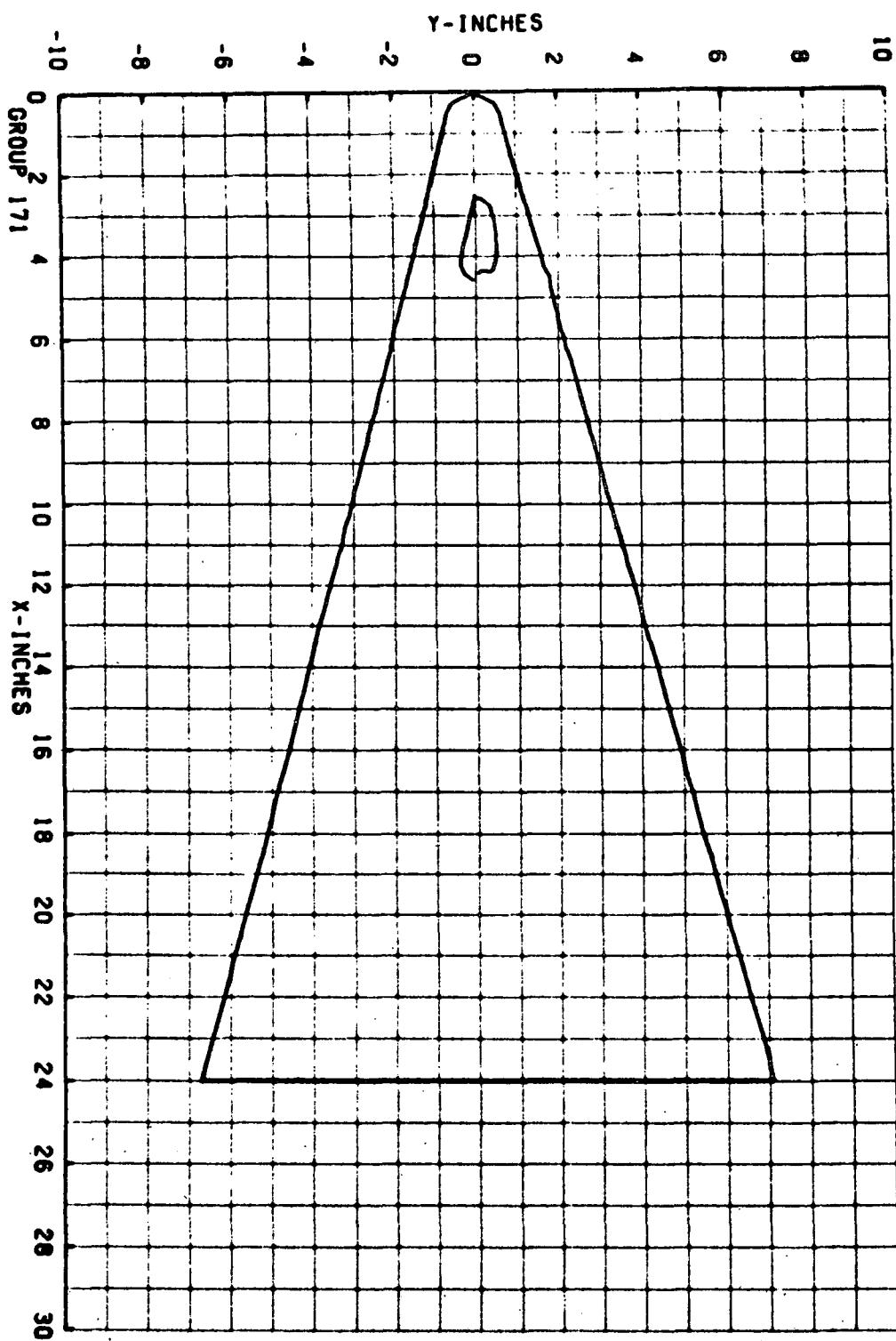
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AEDC (ARO, INC.) ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL B

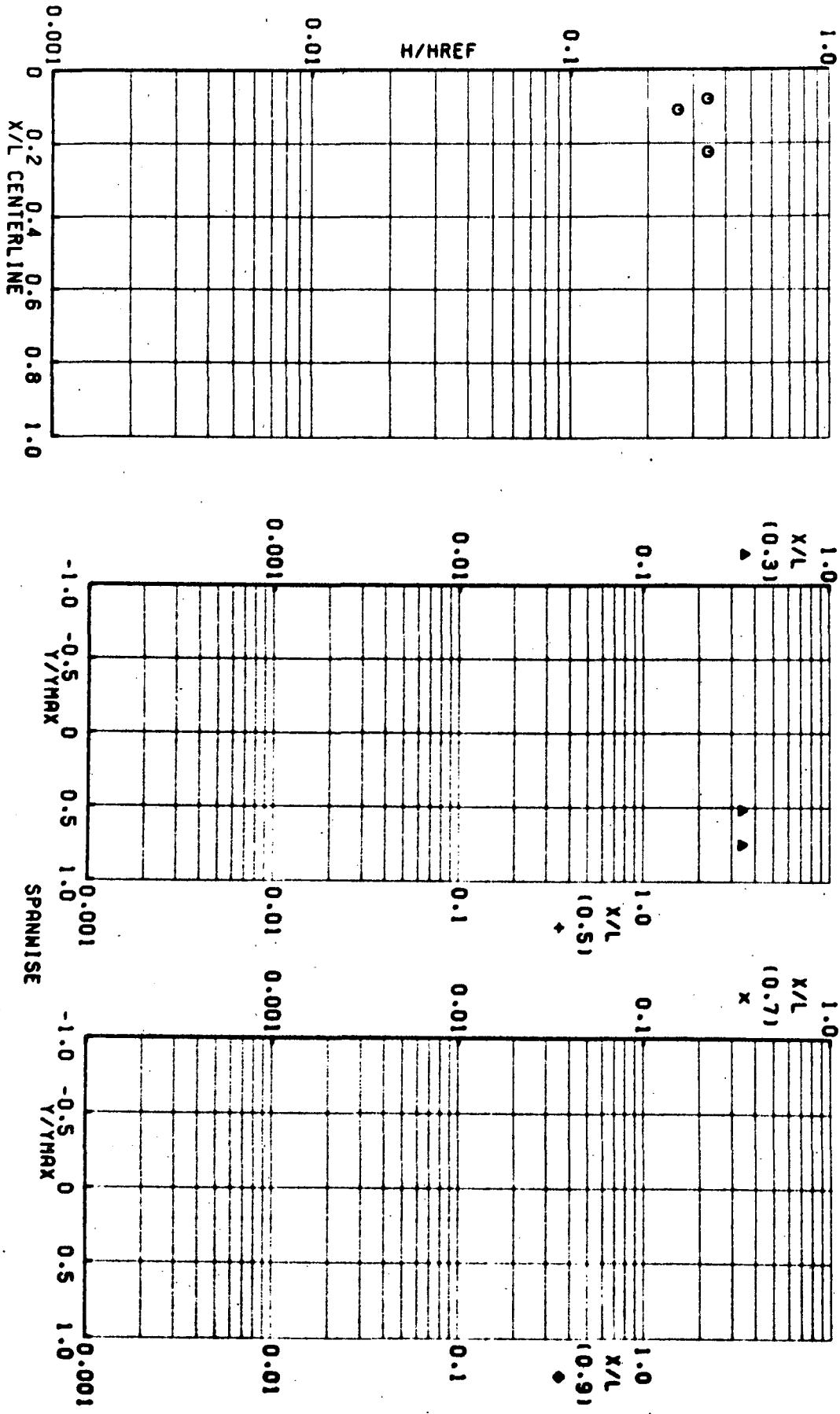


GROUP 171 PIC. NO. 3534 H/HREF 2.602E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.761E-02
RE/FT 3.710E 06 CONF LRC-DB

MODEL SURFACE - BOTTOM



GROUP 171 ALPHA (DEG) 40.0 HREF 2.761E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FT 3.710E 06 CONF LRC-DB



12/11/6

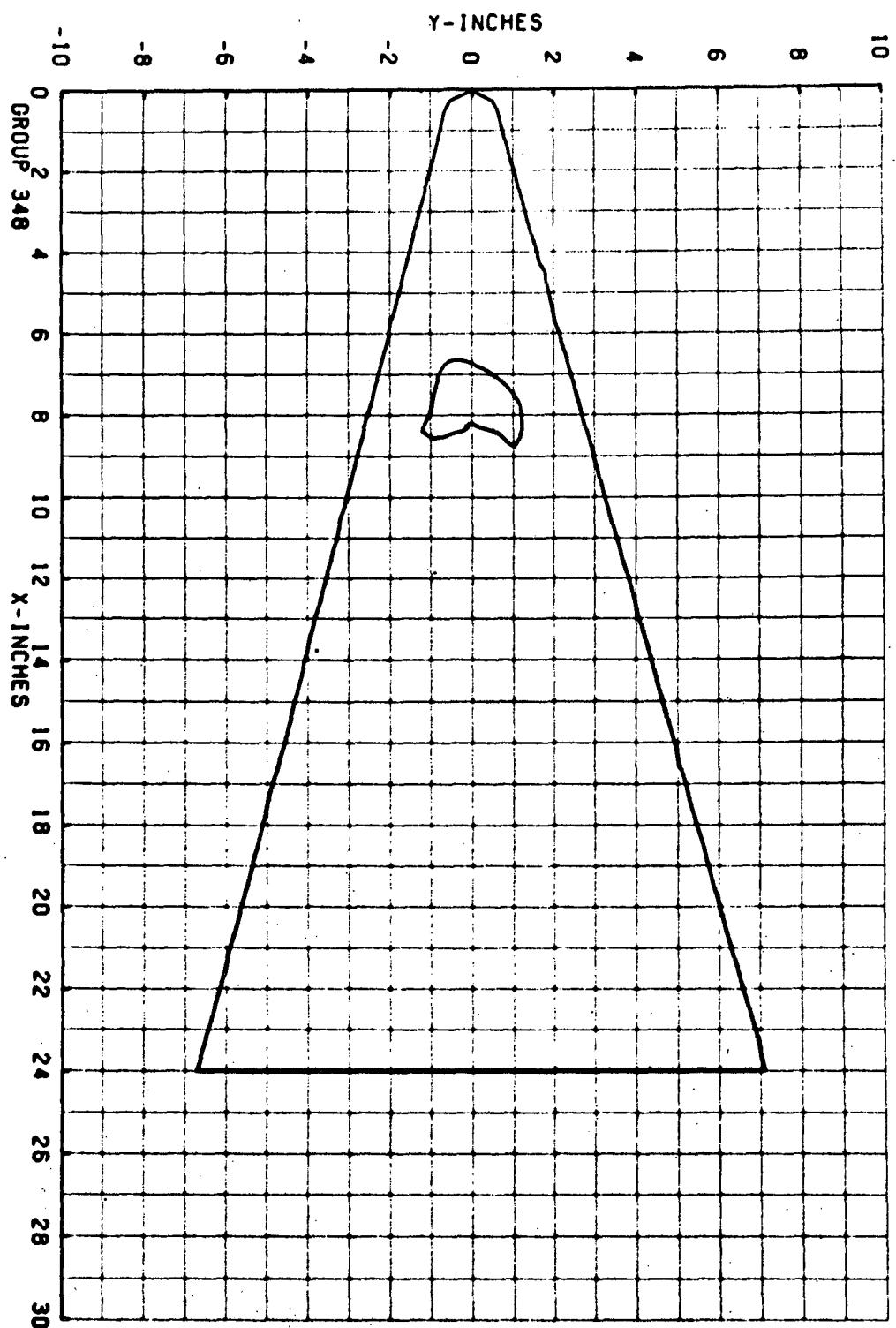
**AEDC (AIAW, INC.) ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY**

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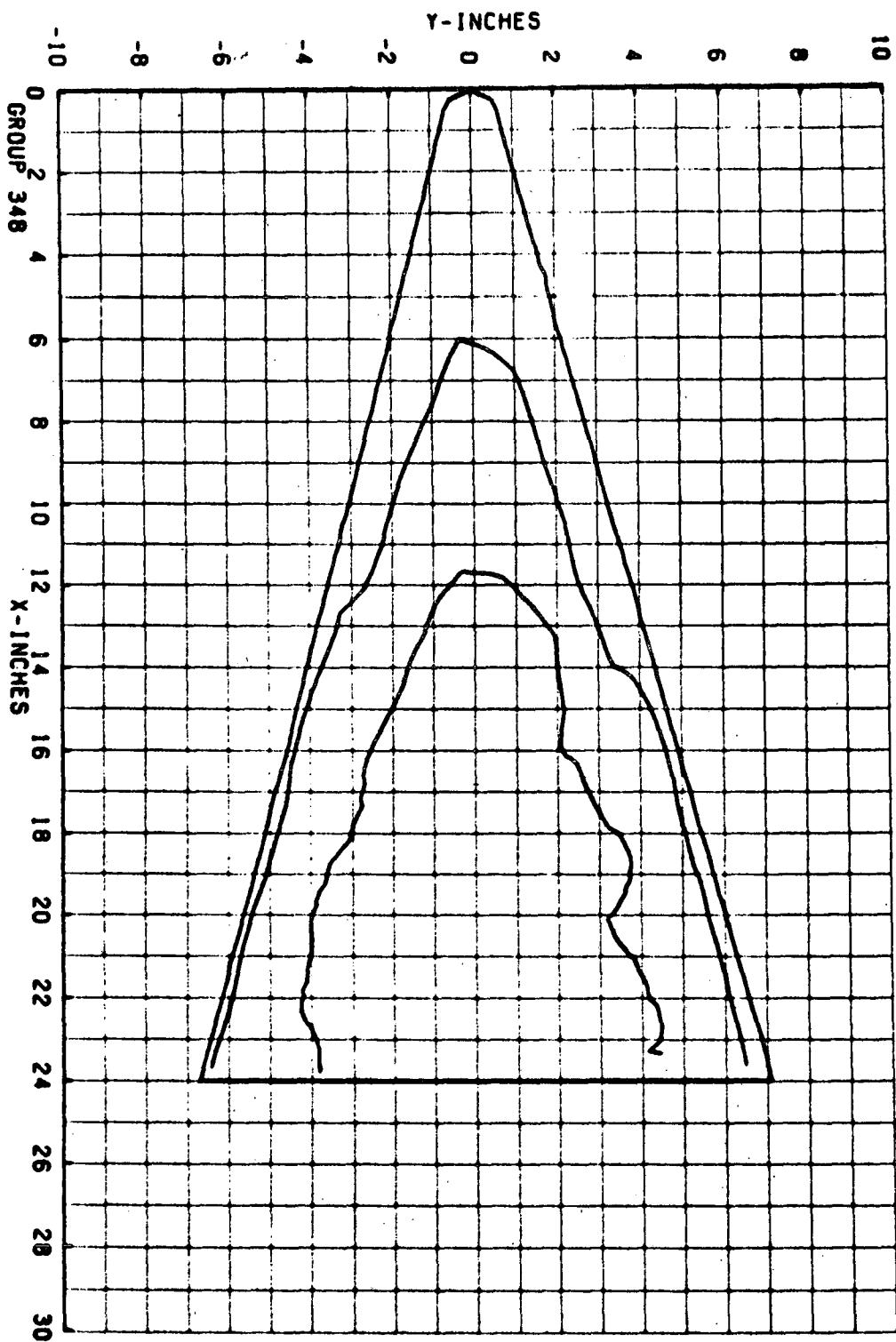
    GROUP CONFIG MODEL MACH NO PROPS TA TO DEG H ALPHA-MODEL ALPHA-SECTOR ALPHA-PREPEND ROLL-MODEL VAN
    348 11 INC-DH 8.00 852.9 1343 40.00 10.00 50.00 180.00 -0.0
    T-INF P-INF Q-INF V-INF RHO-INF MU-INF RE/FT HREF SNEF
    (DEG H) (PSIA) (PSIA) (FT/SEC) (SLUGSF13) (LB-SEC/FT2) (FT-1) (RH= .056FT) (RH= .056FT)
    87.3 1.089 3.974 .3867 7.694E-05 1.834E-08 3.17E-06 2.776E-02 1.166E-02
    CAMERA PAINT TEMP (DEG F) INITIAL TEMP (DEG F) SQUARE ROOT (MMUXCM)
    TOP(1) 400
    SIDE(S) 400 AVERAGE IN 75
    BOTTOM(B) 400

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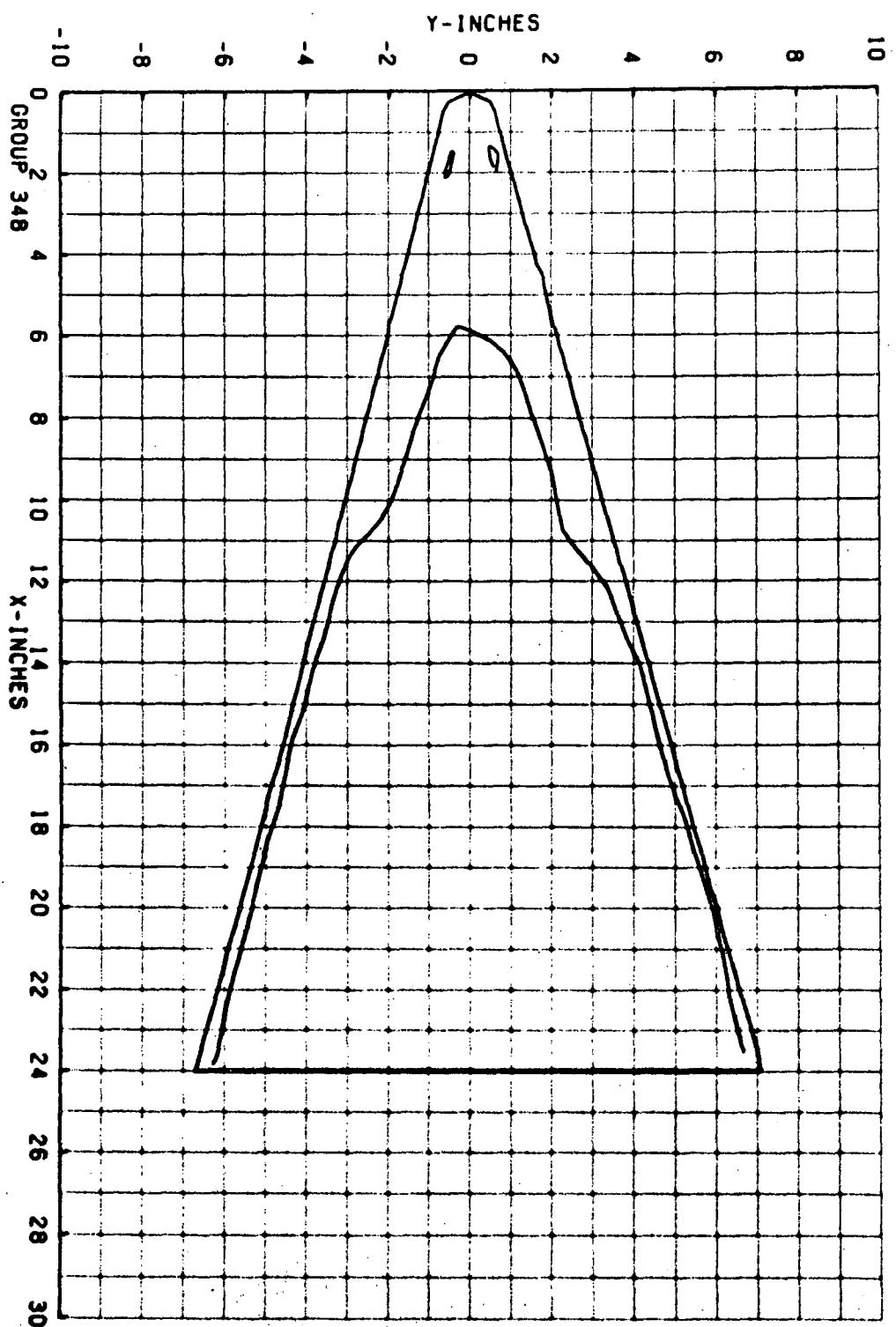
GROUP 348 PIC. NO. 836 H/HREF 5.013E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.776E-02
GROUP 2 348 RE/FT 3.770E 06 CONF LRC-DB

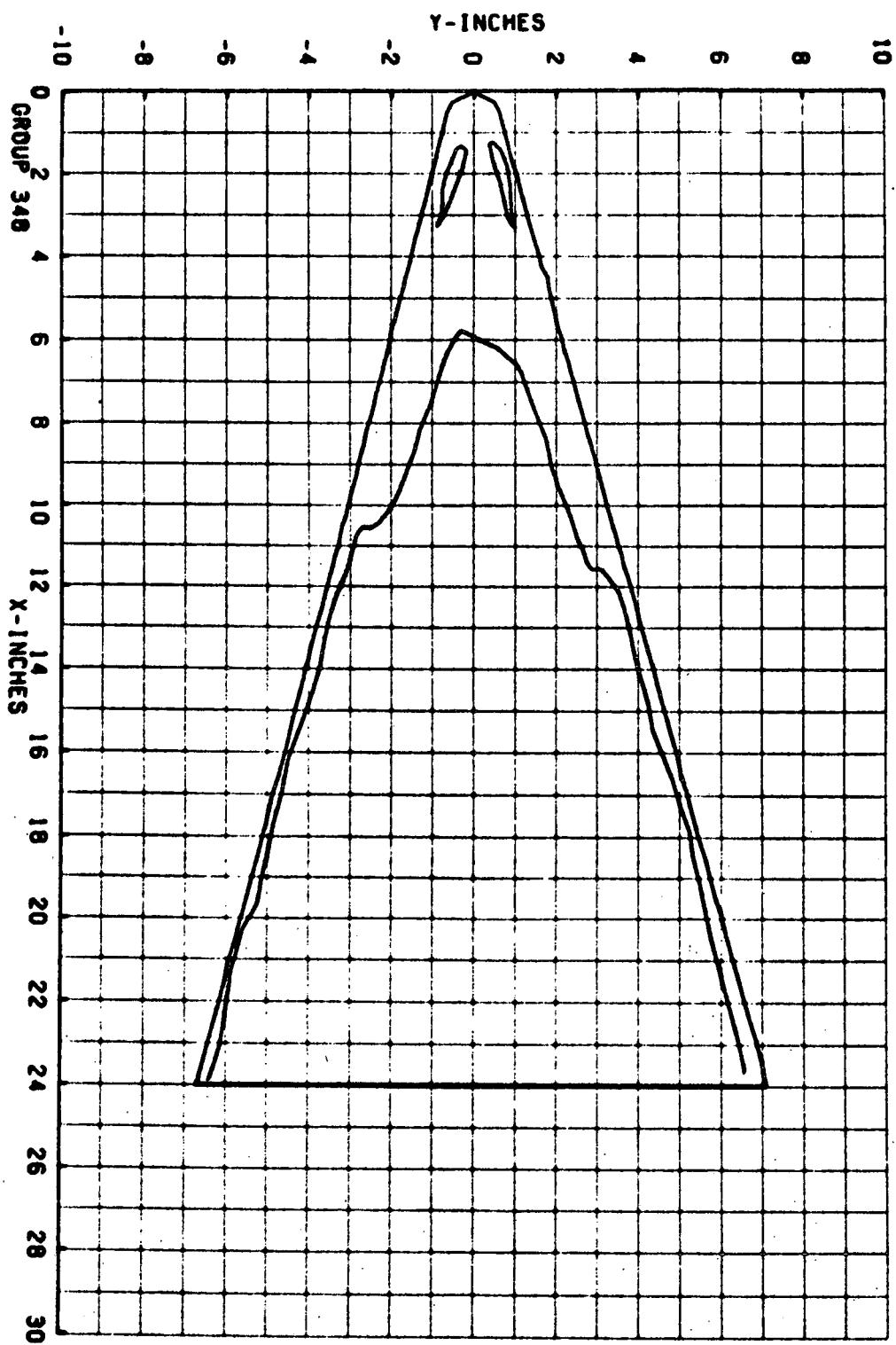


GROUP 348 PIC. NO. 837 H/HREF 4.446E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.776E-02
RE/FT 3.770E 06 CONF LRC-08



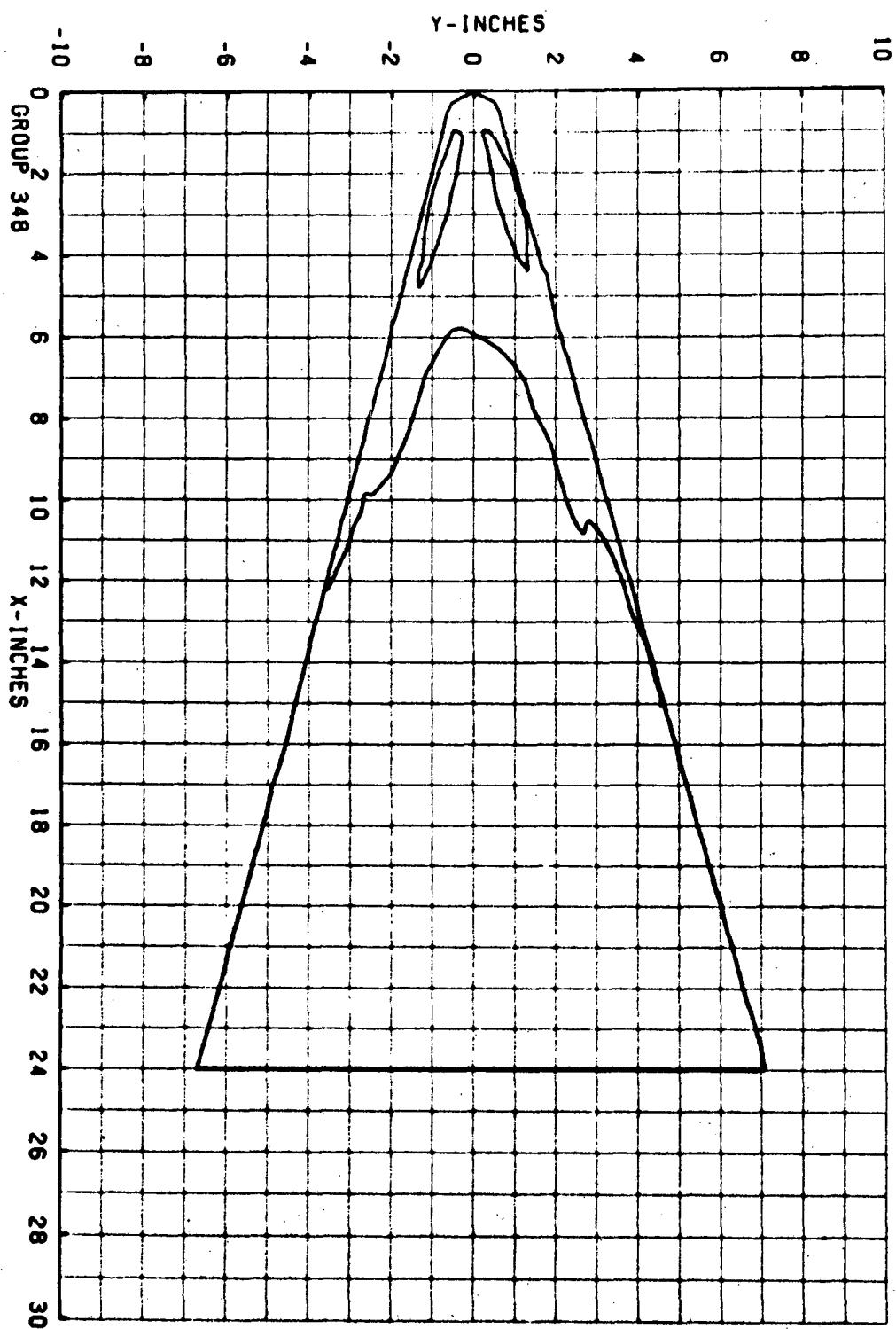
GROUP 348 PIC. NO. 839 H/HREF 3.749E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.776E-02
RE/FT 3.770E 06 CONF LRC-08



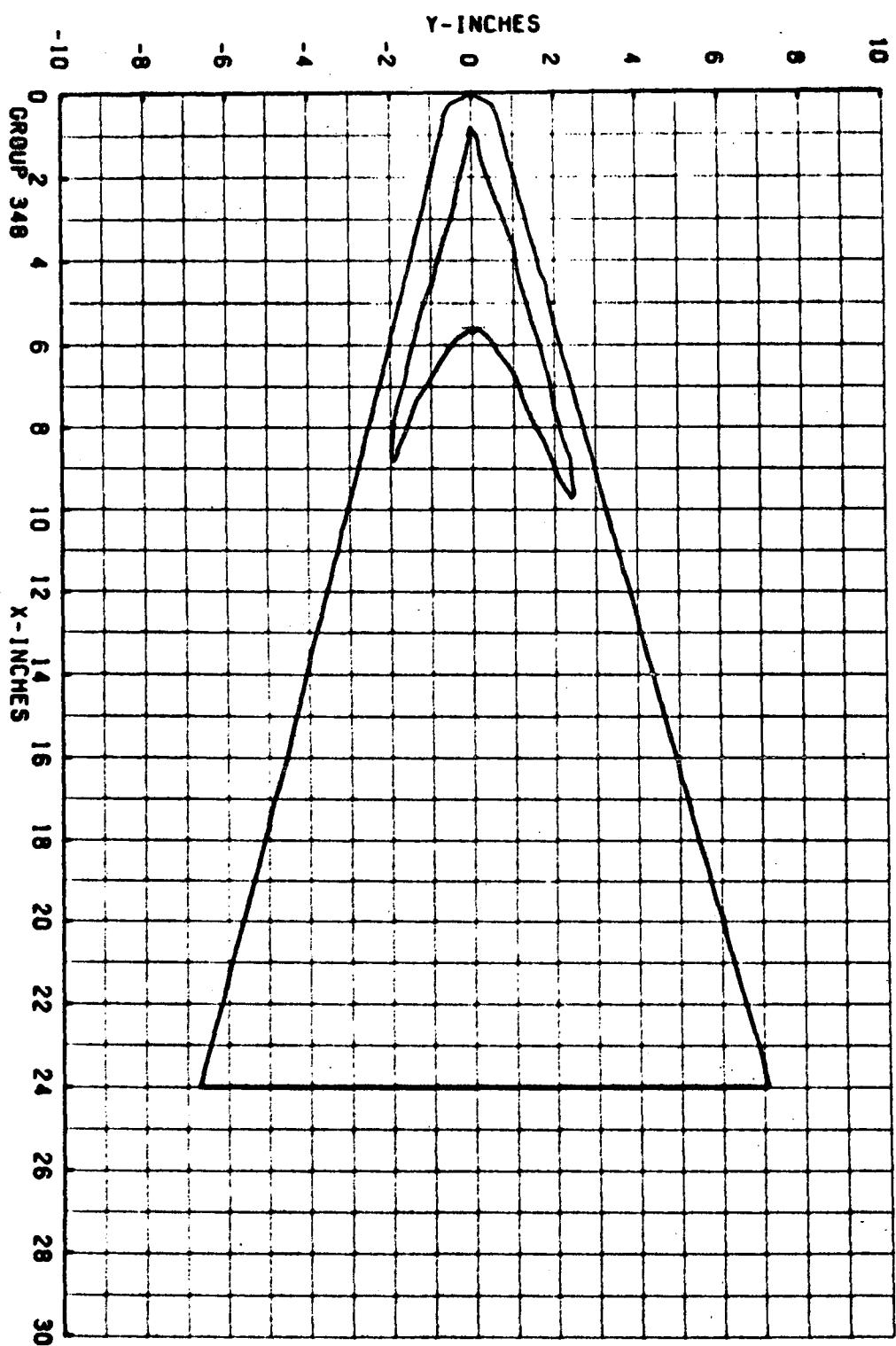


GROUP 348 PIC. NO. 840 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/REF 3.514E-01
ALPHA (DEG) 40.0 HREF 2.776E-02 RE/FT 3.770E 06 CONF LRC-08

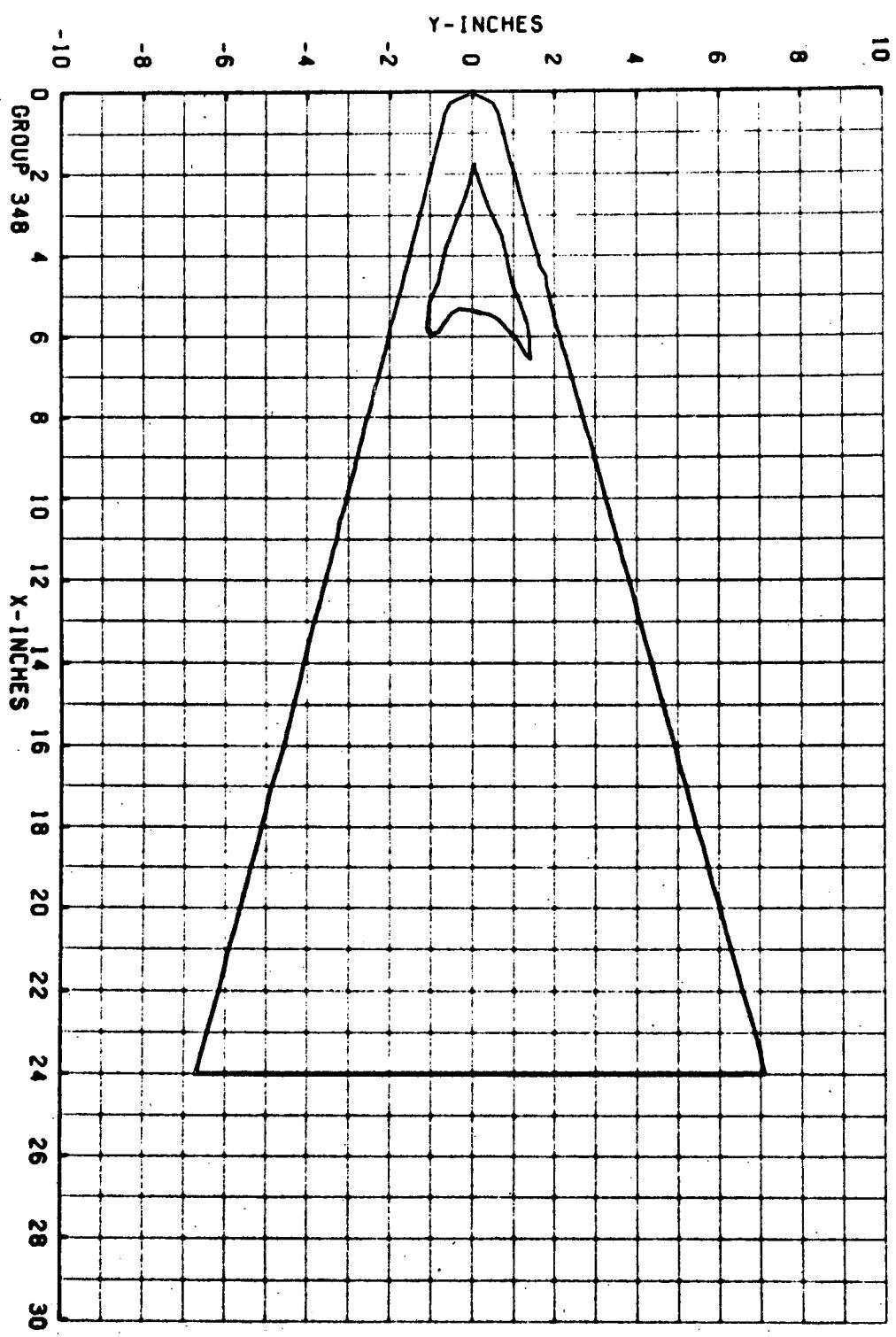
GROUP 348 PIC. NO. 842 H/HREF 3.138E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.776E-02 RE/FT 3.770E 06 CONF LRC-D6



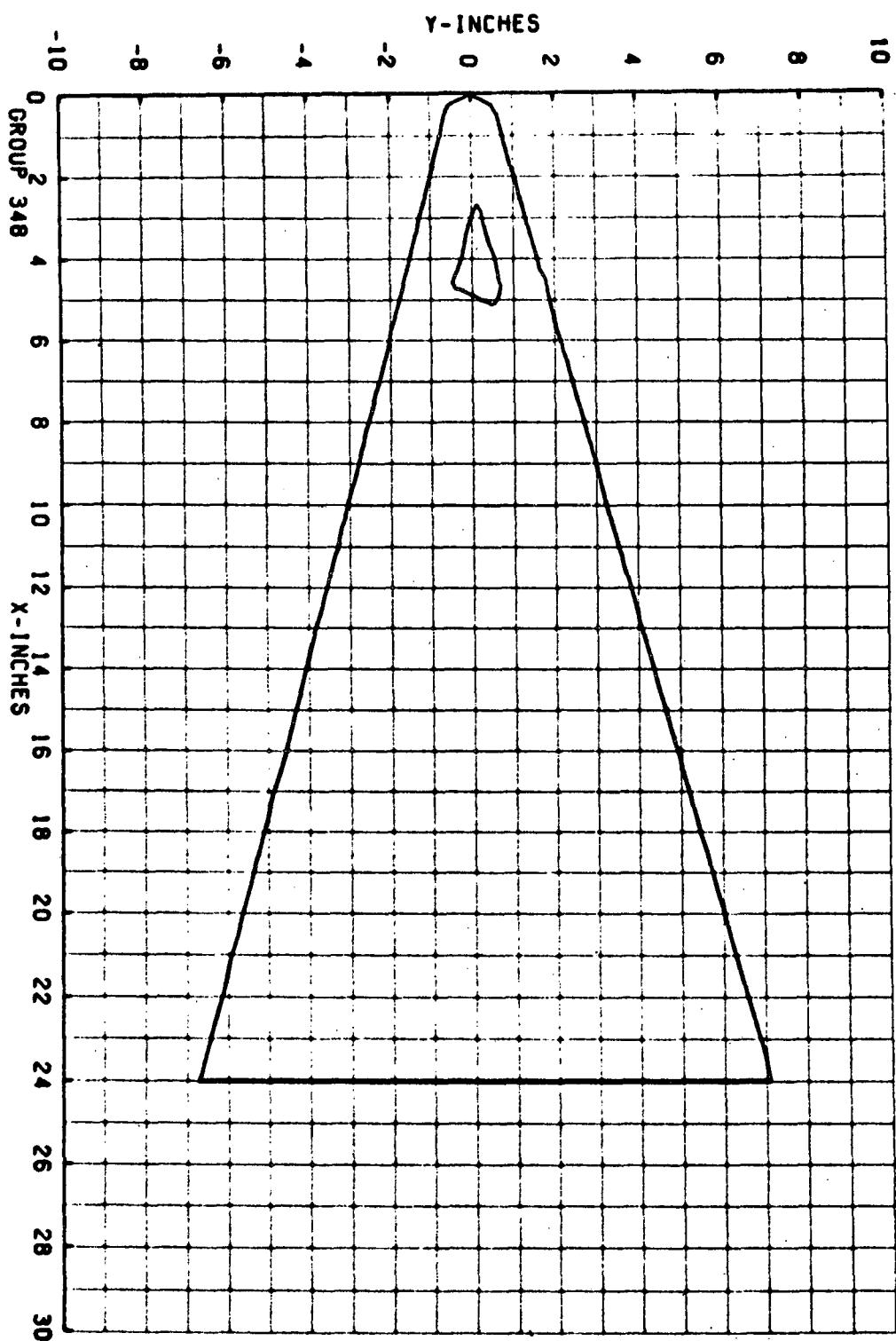
GROUP 348 PIC. NO. 844 H/HREF 2.862E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.776E-02
RE/FT 3.770E 06 CONF LRC-DB
MODEL SURFACE - BOTTOM



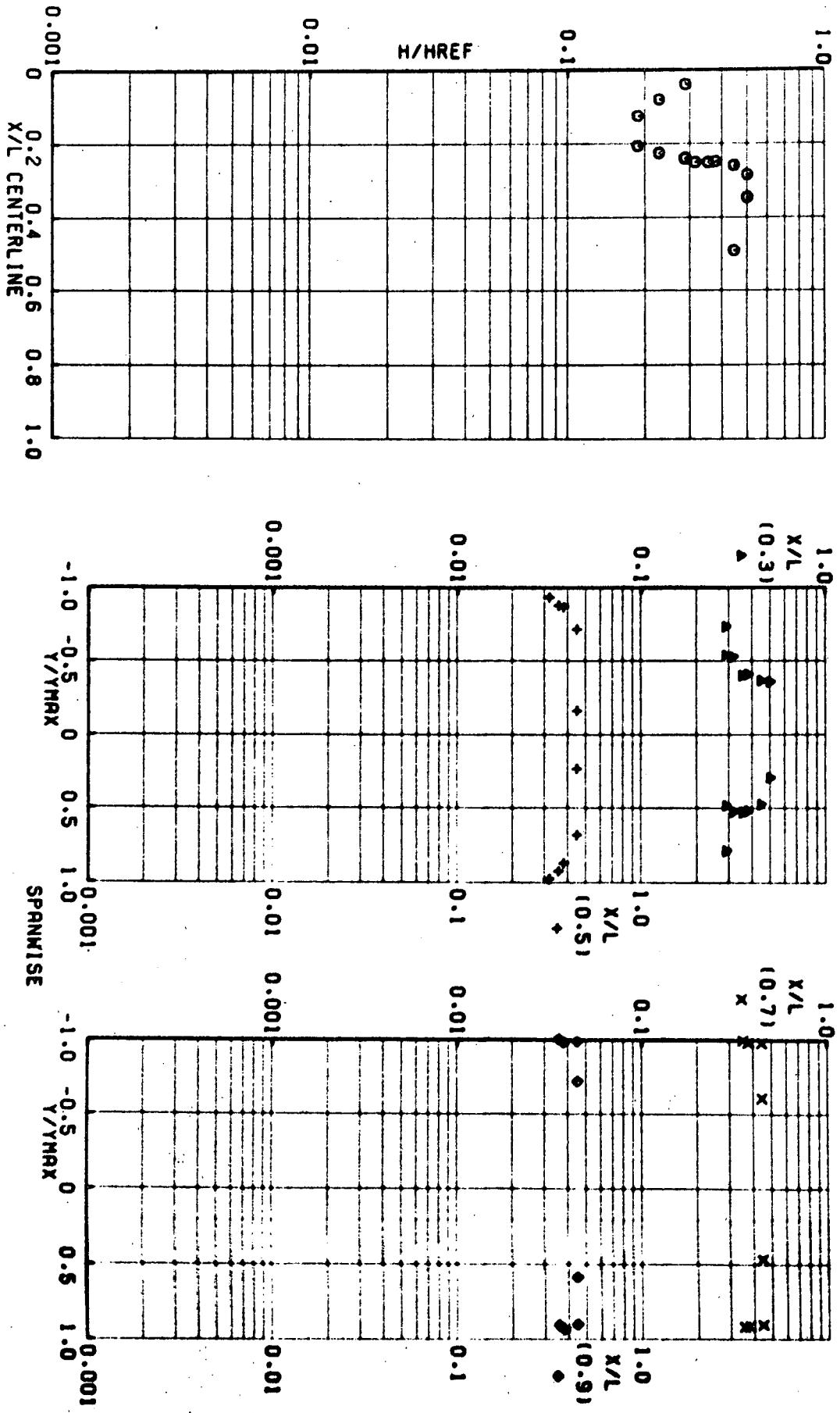
GROUP 348 PIC. NO. 851 M/HREF 2.266E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.776E-02
GROUP 2 348 RE/FIT 3.770E 06 CONF LRC-08
MODEL SURFACE - BOTTOM



GROUP 348 PIC. NO. 860 H/HREF 1.864E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.776E-02 RE/FT 3.770E 06 CONF LRC-DB



GROUP 348 ALPHA (DEG) 40.0 HREF 2.776E-02 MACH 8.00
MODEL SURFACE - BOTTOM RE/FT 3.770E 06 CONF LRC-08



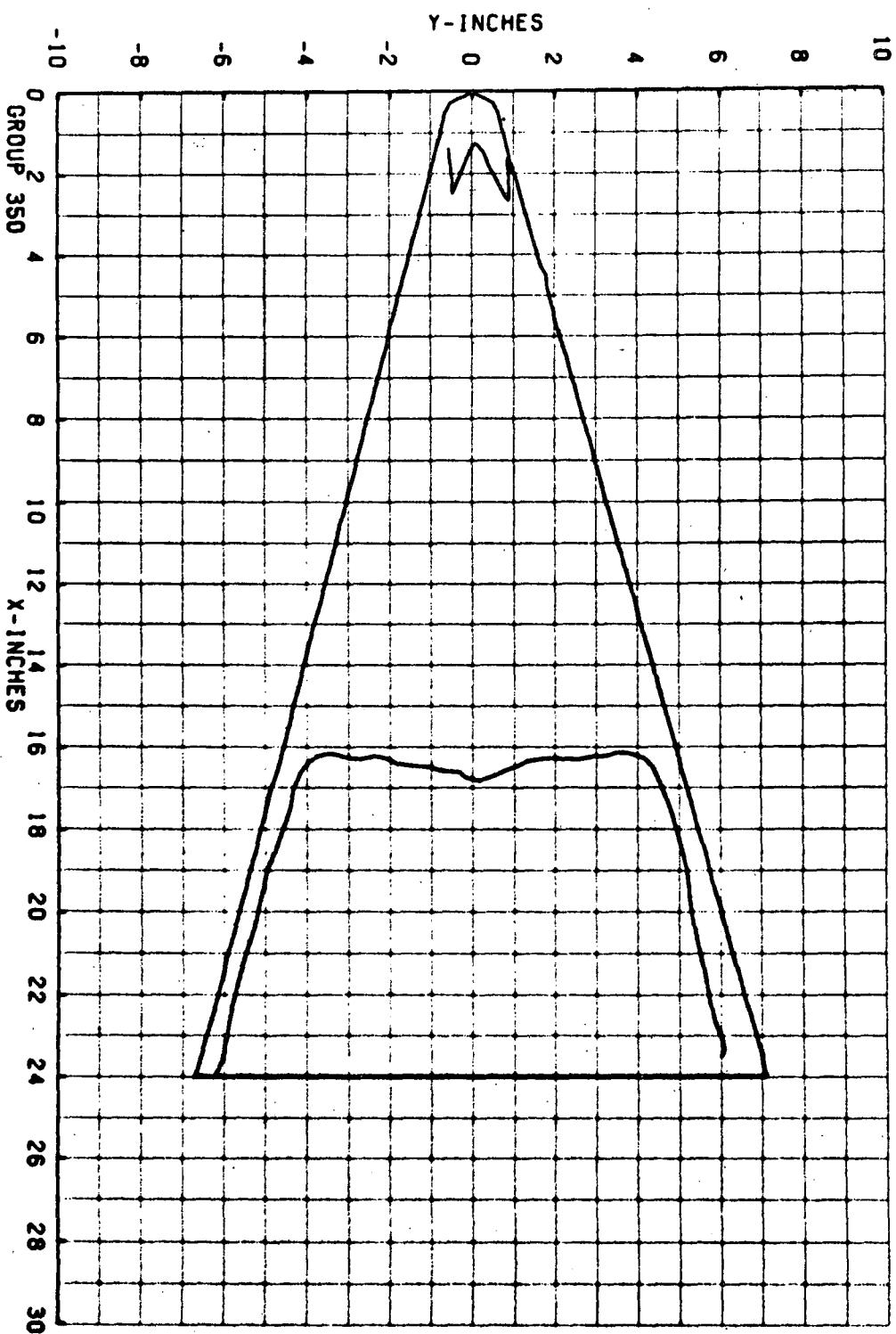
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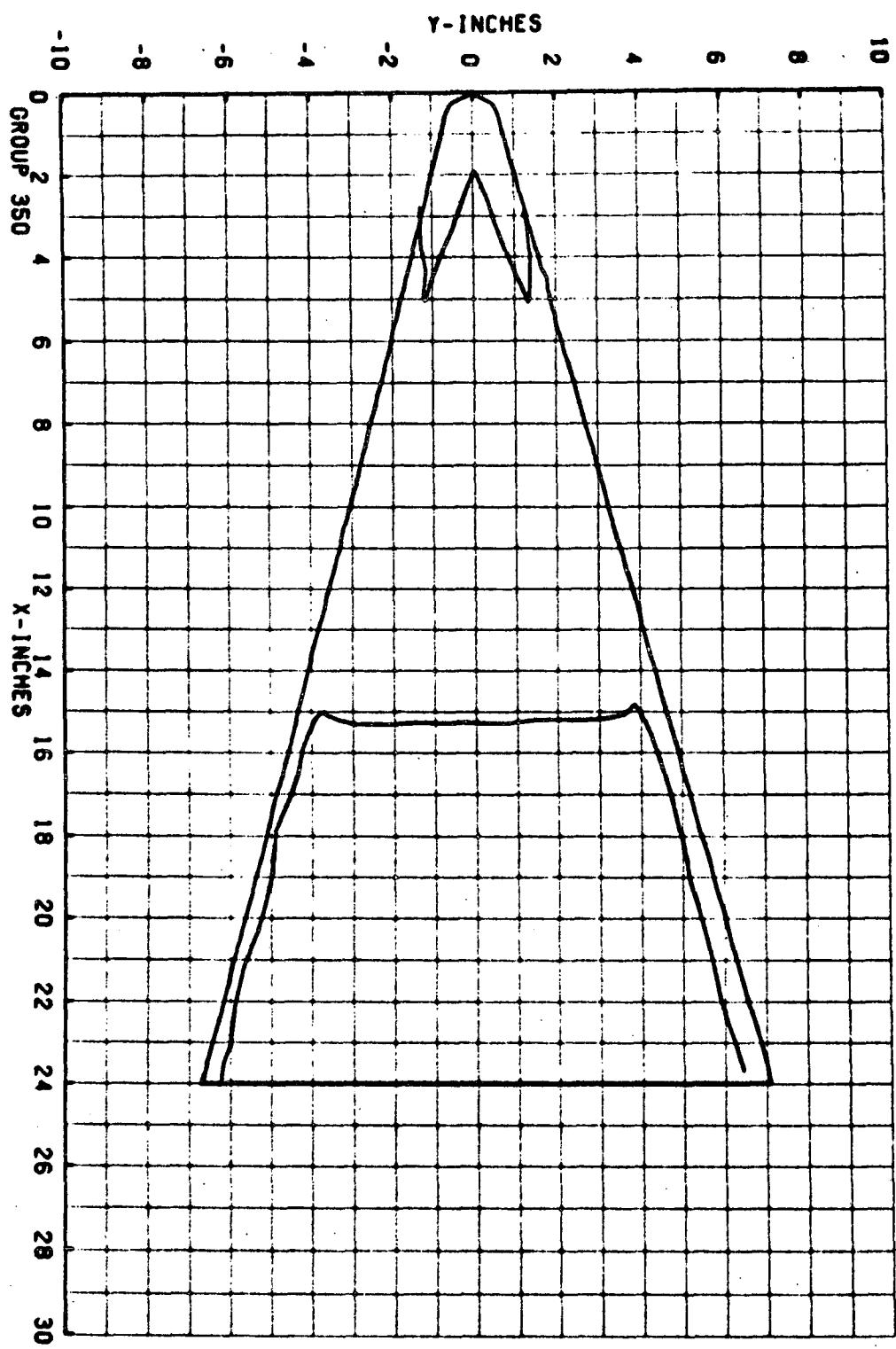
**AEDC(LAHO,INC.) ARNULU AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL B**

GROUP	CONFIG	MODEL	MACH NO	PROPSIA	T0 DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW
350	11	LHC-UH	8.00	861.1	1341	60.00	-10.00	50.00	180.00	-.0
I-INF	P-INF	Q-INF	V-INF	RHO-INF	MU-INF	REFIT	MREF	SREF		
(UEG F)	(PSIA)	(PSIA)	(FT/SEC)	(SLUGS/FT ³)	(LB-SEC/FT ²)	(FT-1)	(R= .056FT)	(R= .056FT)		
97.2	.0148	3.952	.3864	7.616E-19	7.853E-08	3.79E-06	2.167E-02	1.169E-02		
CAMERA	PAINT	TEMP (UEG F)	INITIAL (TEMP (UEG F))	SQUARE ROOT (MMOXCM)						

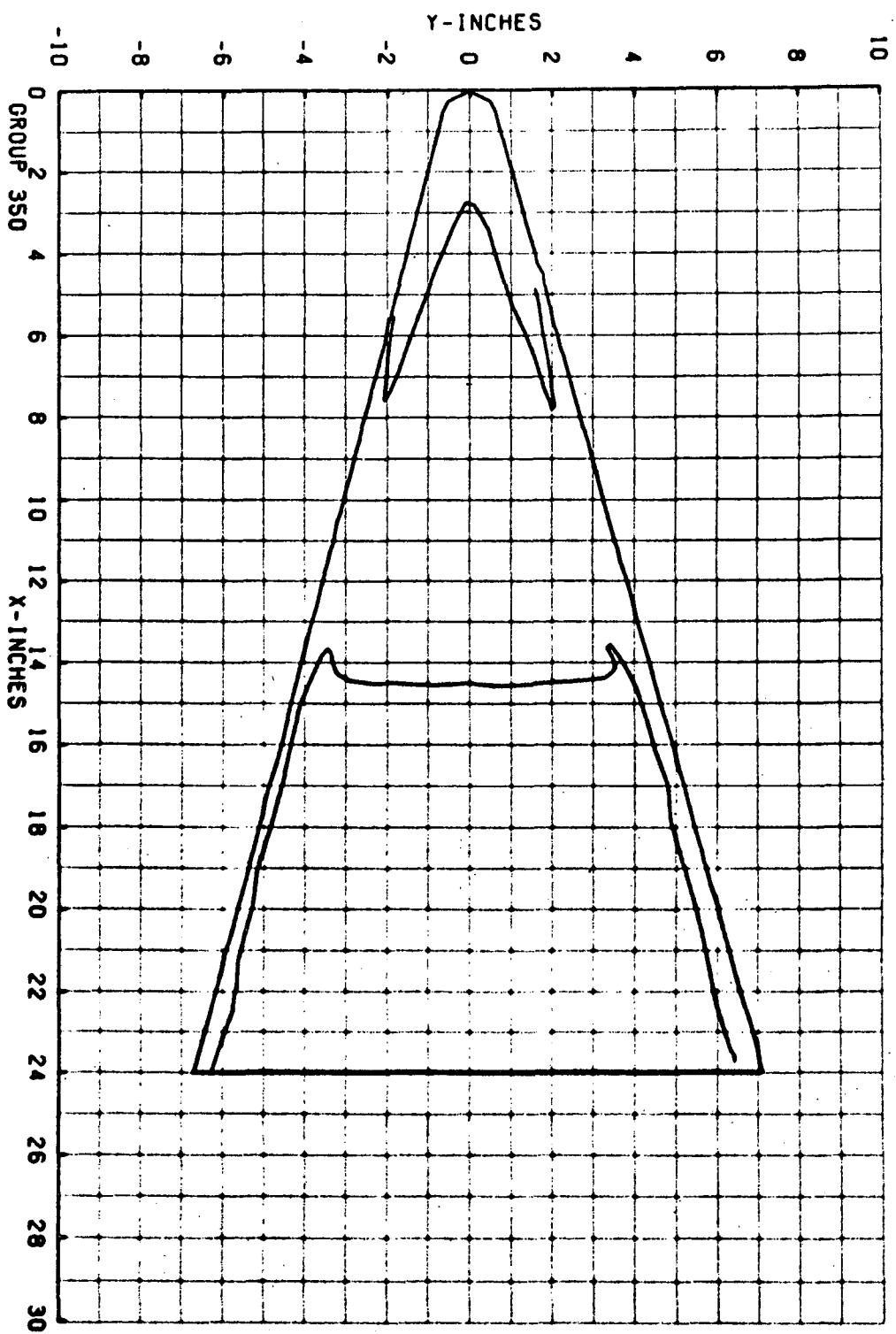
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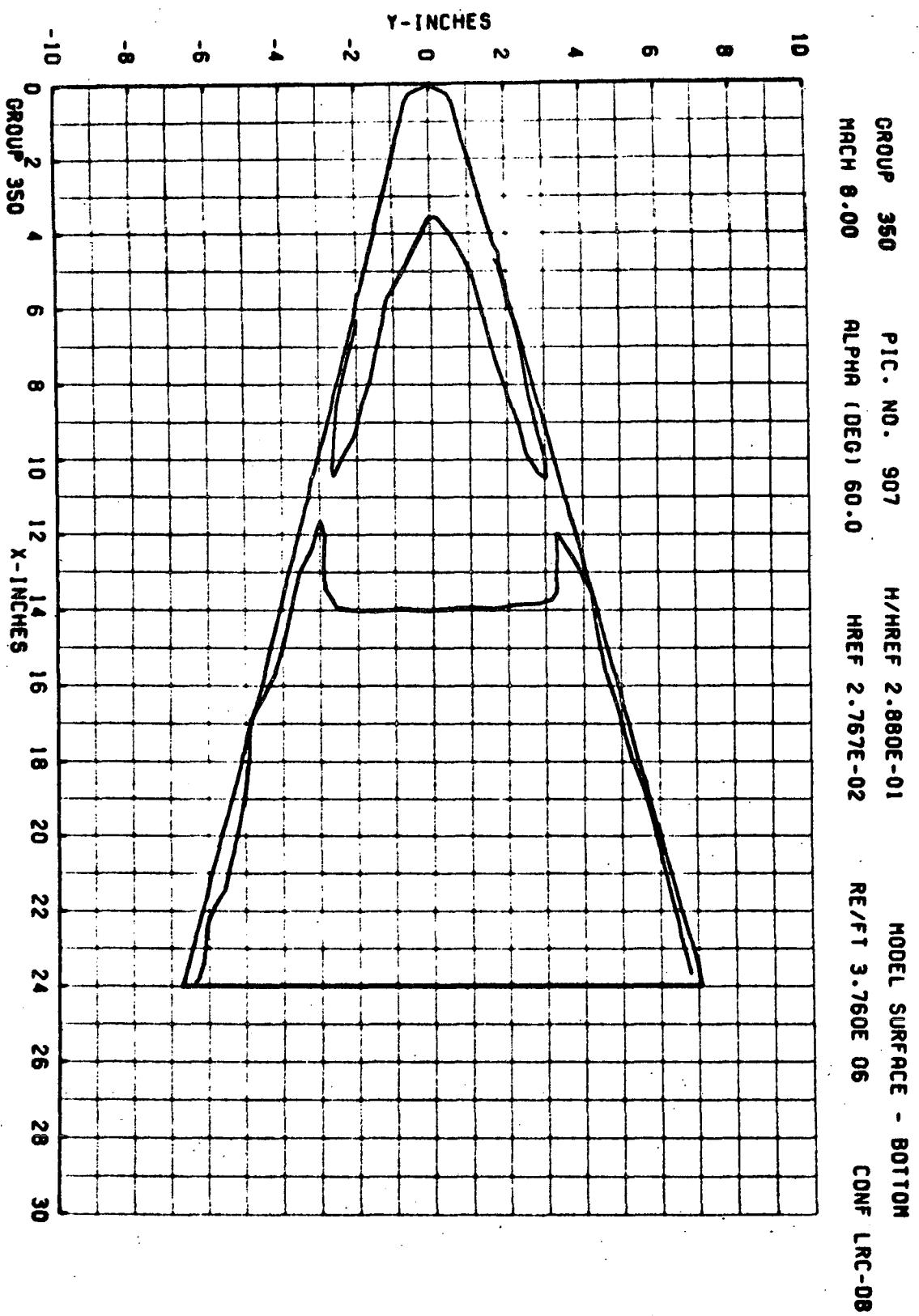
GROUP 350 PIC. NO. 901 H/HREF 4.307E-01
MACH 8.00 ALPHA (DEC) 60.0 HREF 2.767E-02
REF/FT 3.760E 06 CONF LRC-D6

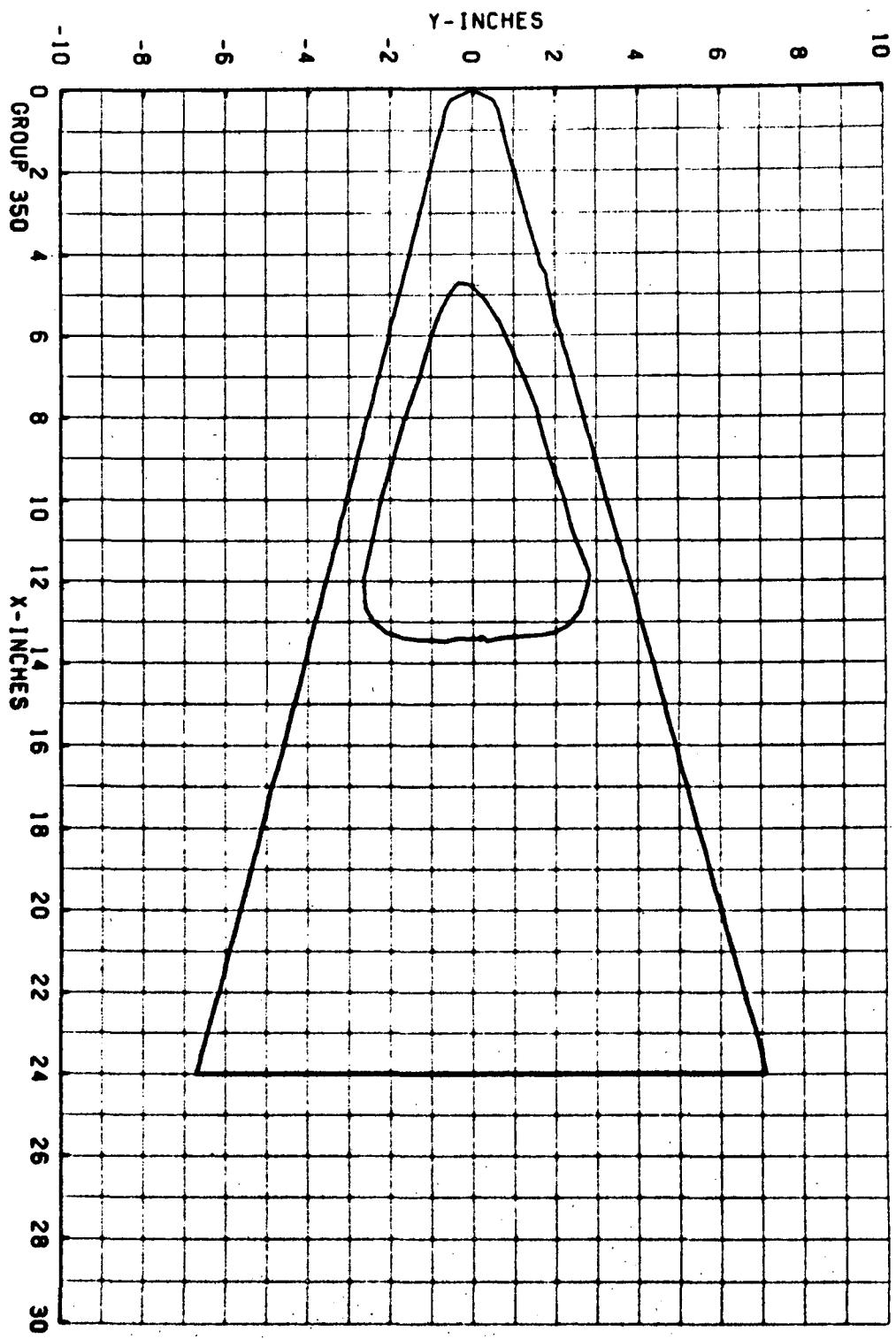


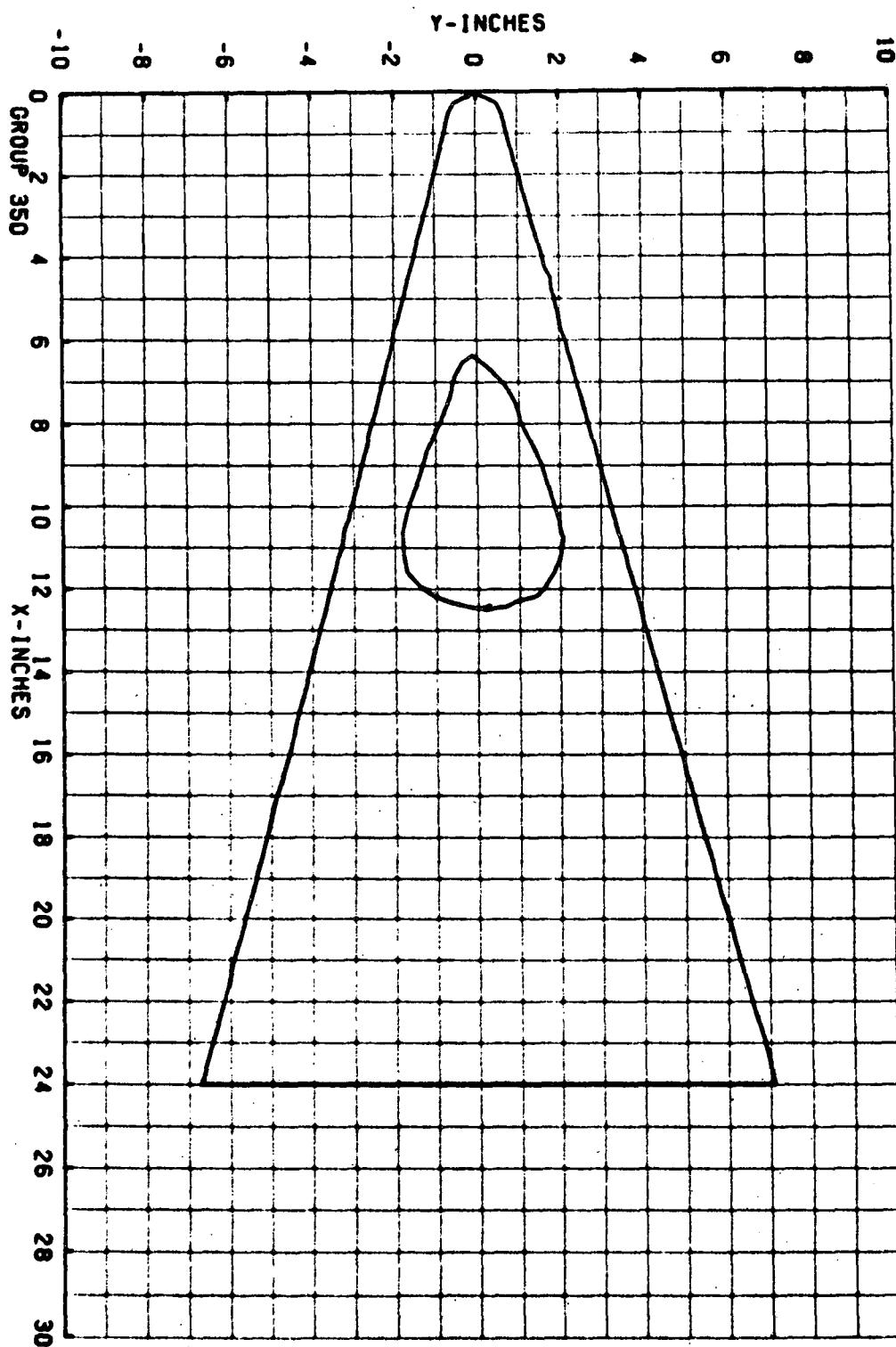


GROUP 350 PIC. NO. 905 H/HREF 3.190E-01
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.767E-02
RE/FT 3.760E 06 CONF LRC-08



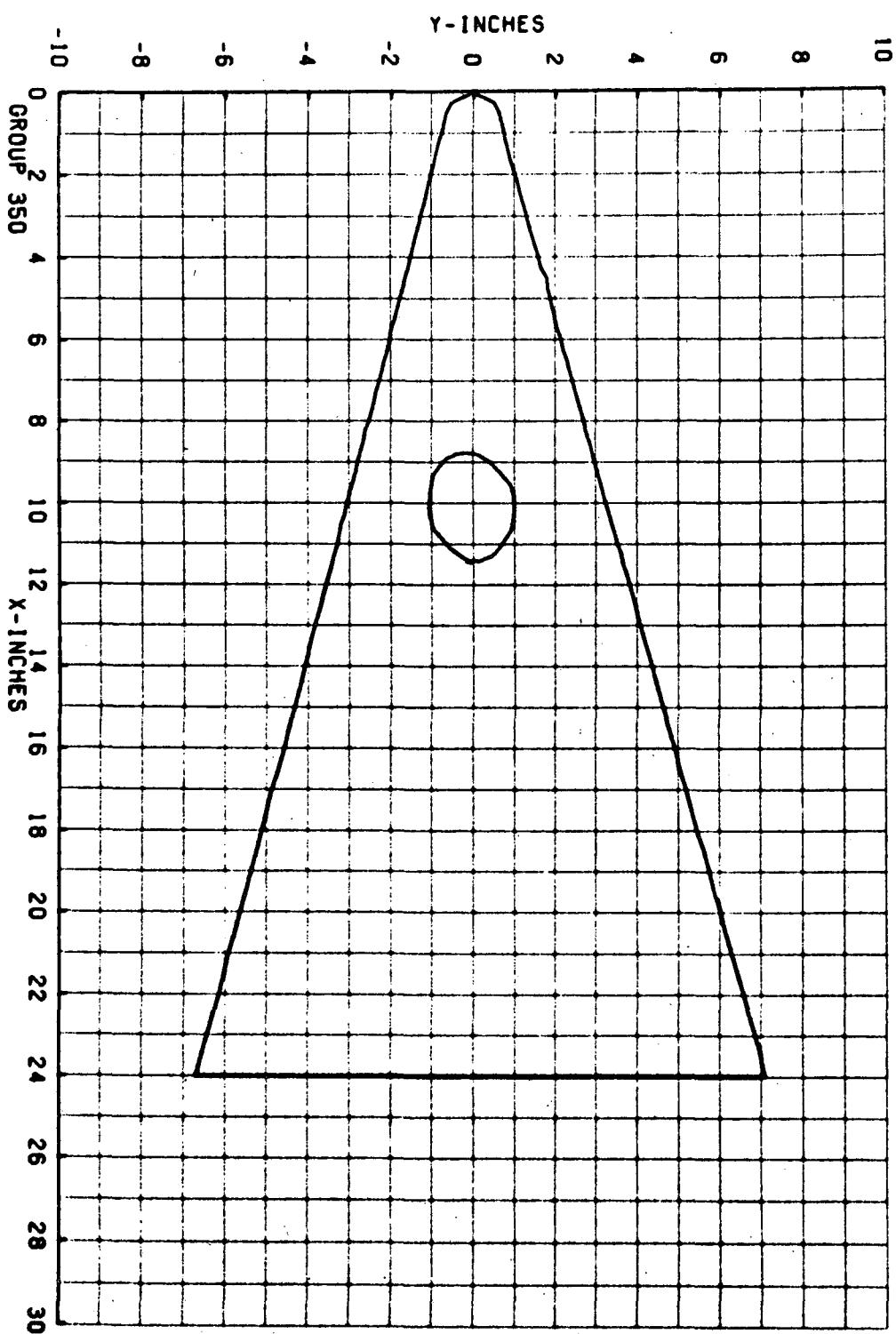




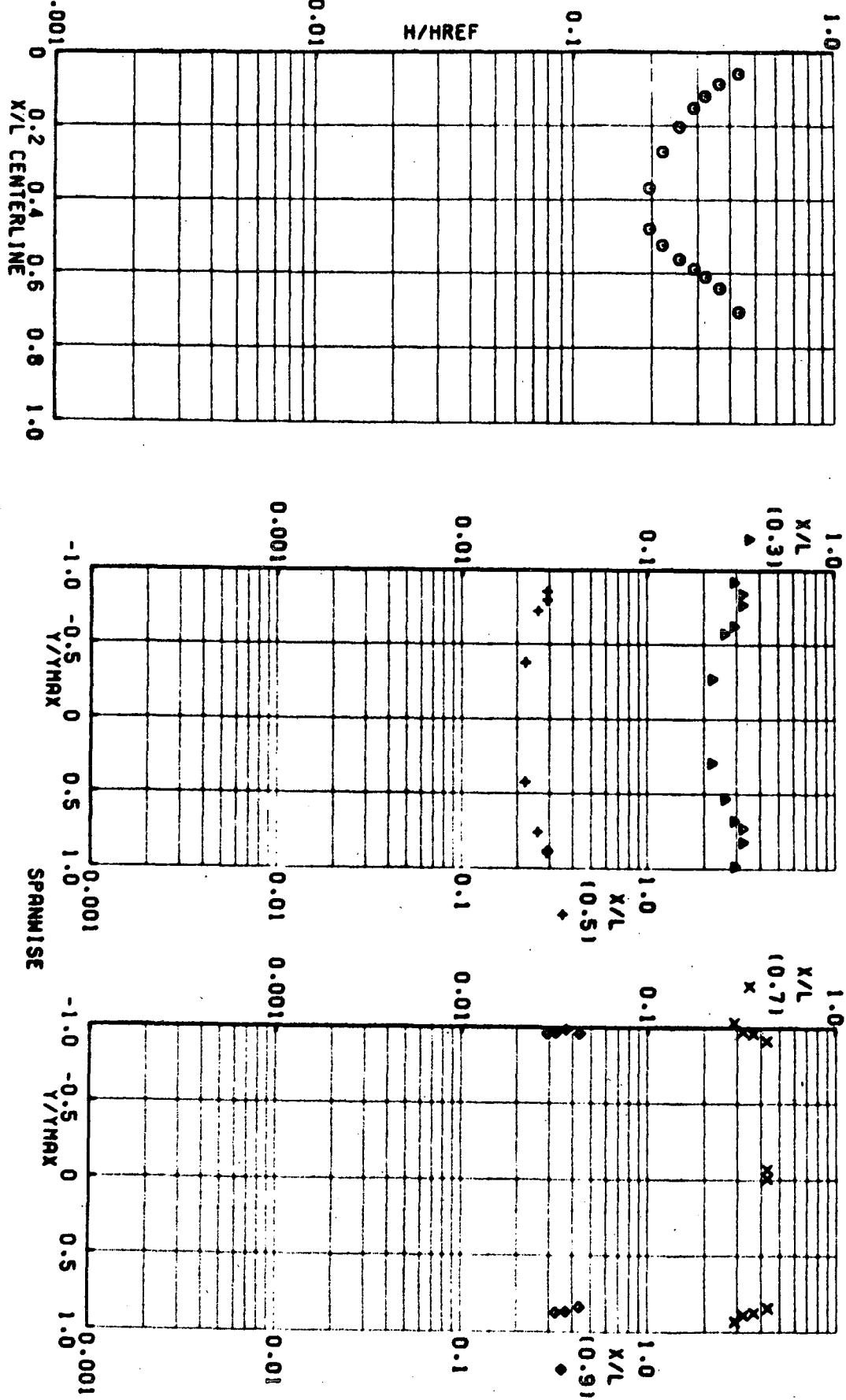


GROUP 350 PIC. NO. 915 H/HREF 2.180E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.767E-02 RE/FT 3.760E 06 CONF LRC-08

GROUP 350 PIC. NO. 920 H/HREF 1.941E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.767E-02 RE/FT 3.760E 06 CONF LRC-08



GROUP 350 ALPHA (DEG) 60.0 MREF 2.767E-02
MODEL SURFACE - BOTTOM RE/F1 3.760E 06 CONF LRC-06
MACH 8.00

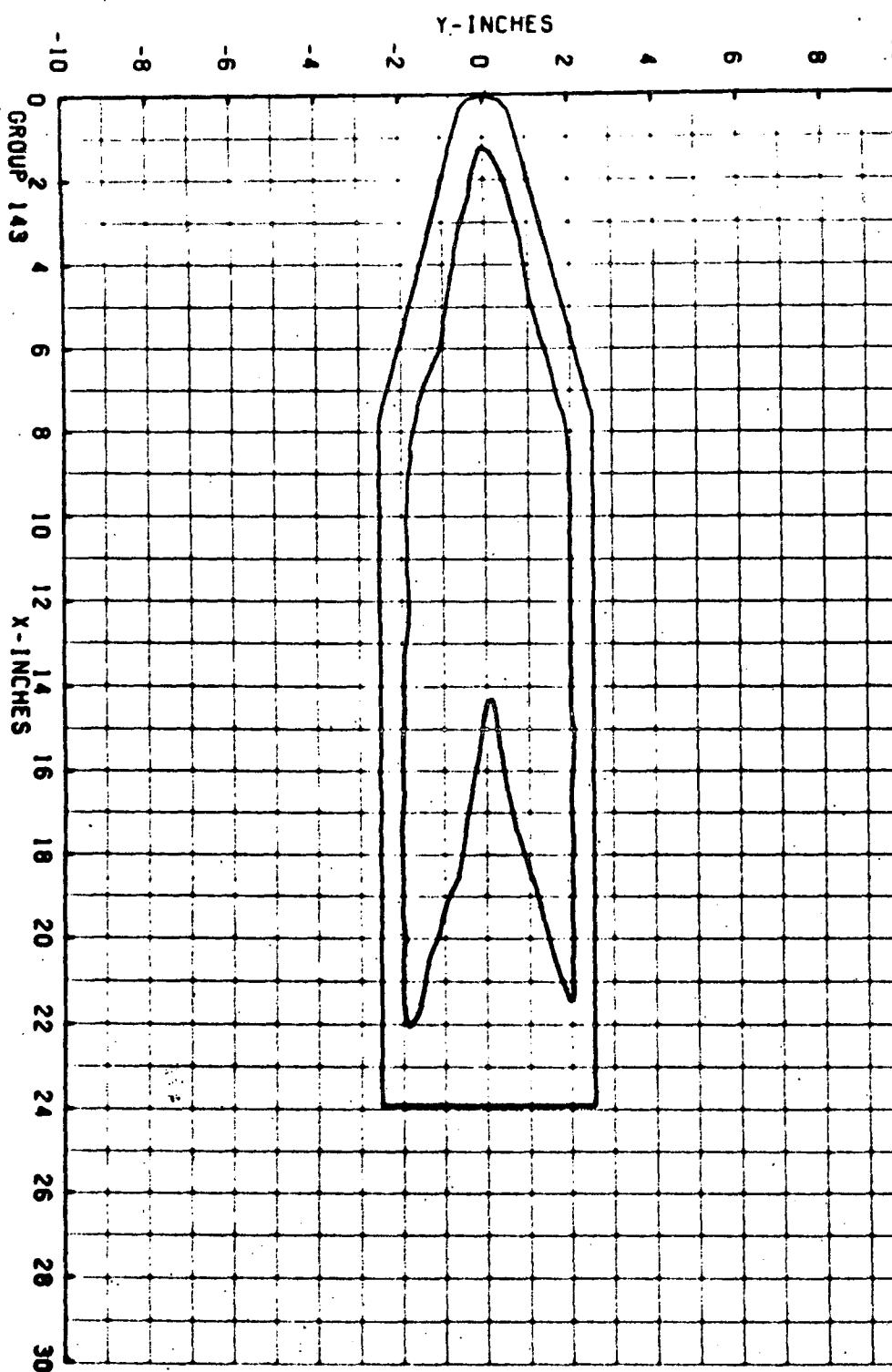


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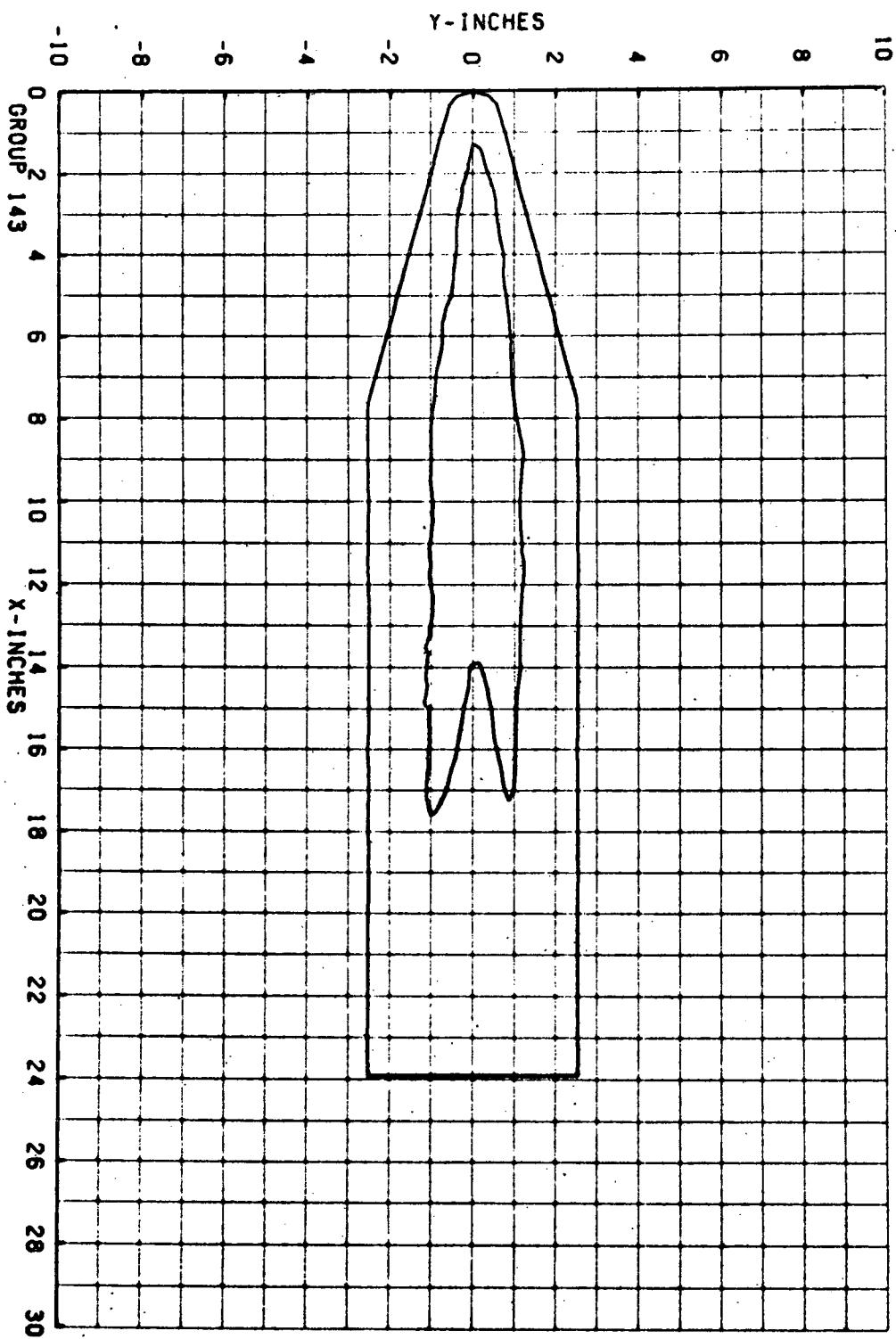
AEDC/LARO, INC.) ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL A
VII-12

GROUP	CONFIG	MODEL	MACH NO	PIN PSIA	TO DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW
1.3	12	IRC-SR	8.00	554.8	1306	19.99	3.01	-23.00	180.00	.0
T-INF	P-INF	Q-INF	V-INF	RHO-INF	MU-INF	REF	HREF	STREF		
(NEG R)	(PSIA)	(PSIA)	(FT/SEC)	(SLUGS/FT ³)	(LB-SEC/FT ²)	(FT-1)	(R= .056FT)	(R= .056FT)		
94.6	.657	2.546	3A13	5.038E-05	7.619E-08	2.52E 06	2.211E-02	1.433E-02		
CAMERA	PAINT	TRIP (DEG F)	INITIAL TEMP (DEG F)	SQUARE ROOT (RHOXCXK)						
TOP (T)	12E									
SIDE (S)	125	AVERAGE T _W =	79	-0.008(SQUARE ROOT OFL TIME) + 0.11						
ROT CM (R)	12E									
PIC NC	TIME DELTIME	H(10)	H(10)/HREF	H(.970)	H(.970)/HREF	H(.8570)	H(.8570)/HREF	ST(10)	MODEL TEMP F	

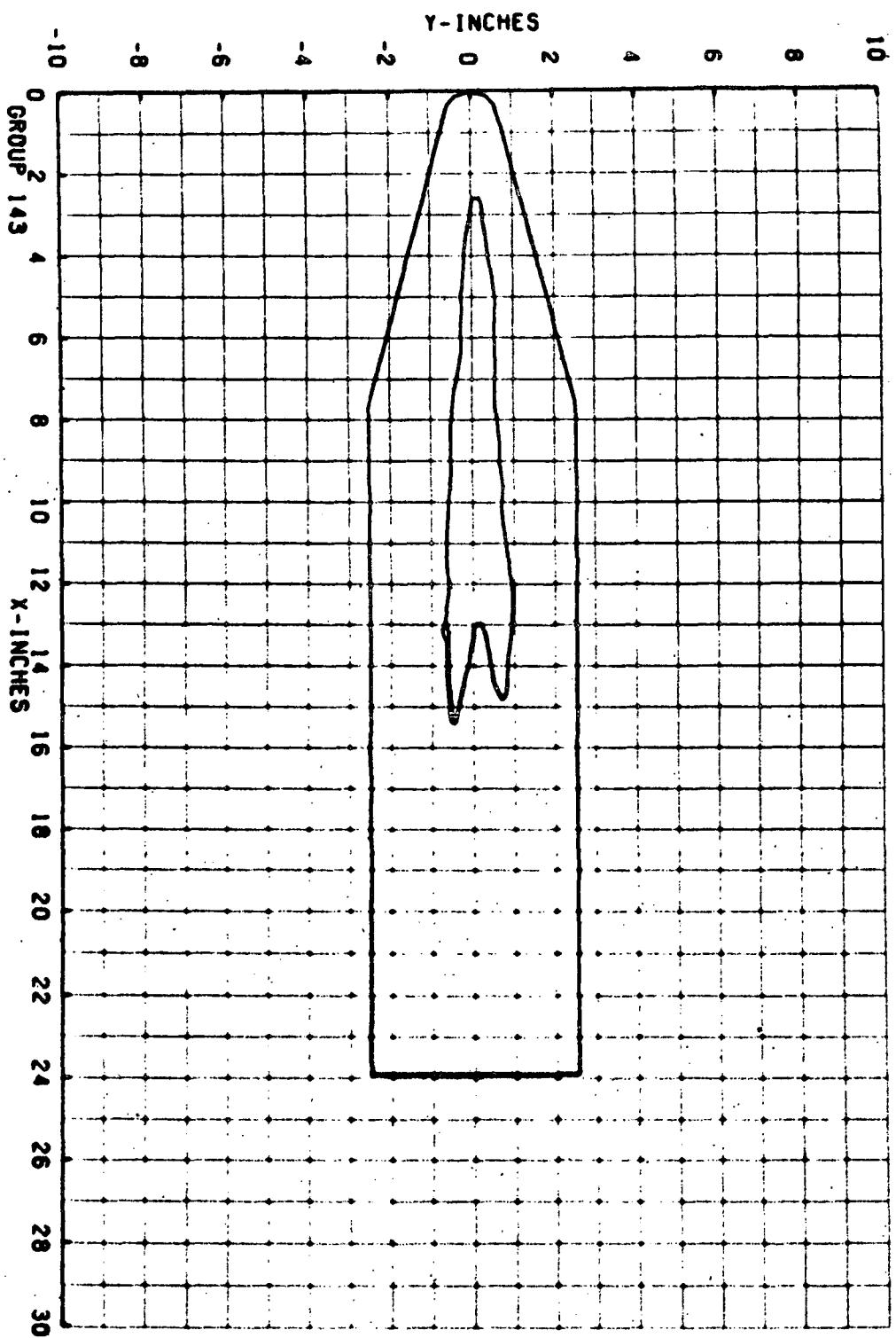
GROUP 143 PIC. NO. 2414 H/HREF 1.992E-01
MACH 8.00 ALPHA (DEC) 20.0 HREF 2.211E-02
RE/F1 2.520E 06 CONF LRC-SB



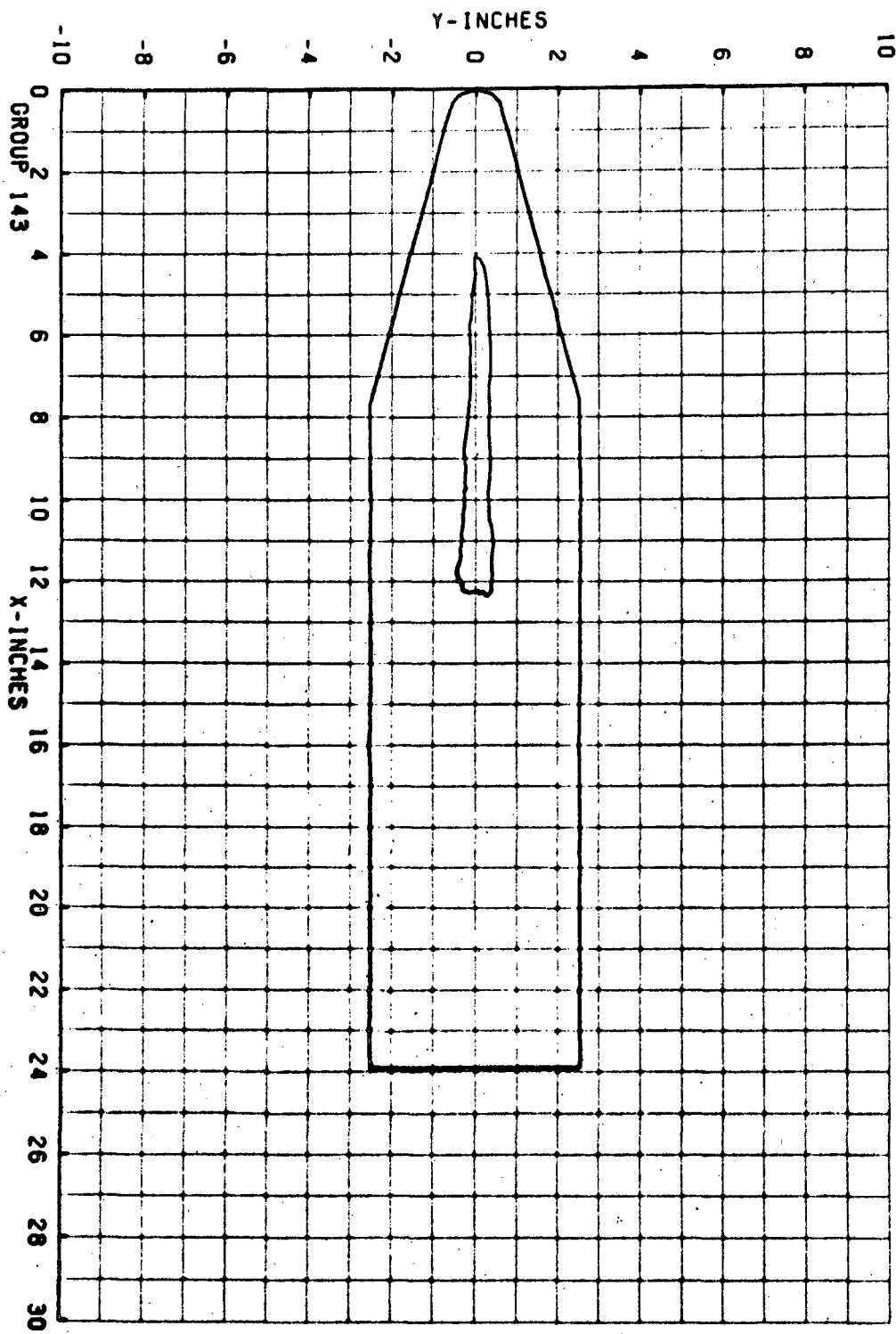
GROUP 143 PIC. NO. 2416 MODEL SURFACE - BOTTOM
MACH 8.00 H/HREF 1.505E-01
ALPHA (DEC) 20.0 HREF 2.211E-02
HREF 2.520E-06 CONF LRC-S8



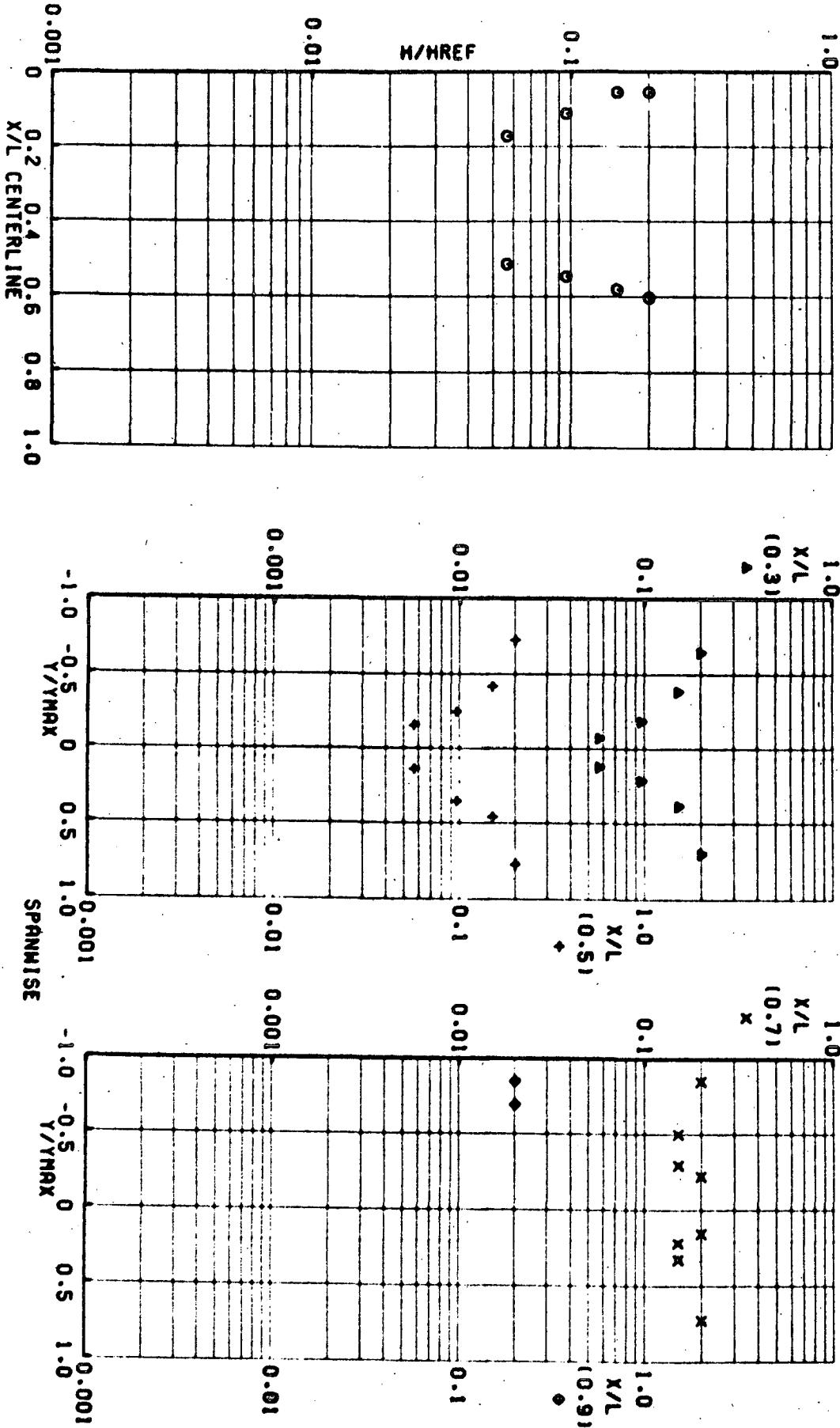
GROUP 143 PIC. NO. 2422 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 20.0 H/REF 2.211E-02
GROUP 143 RE/FT 2.520E 06 CONF LRC-S8



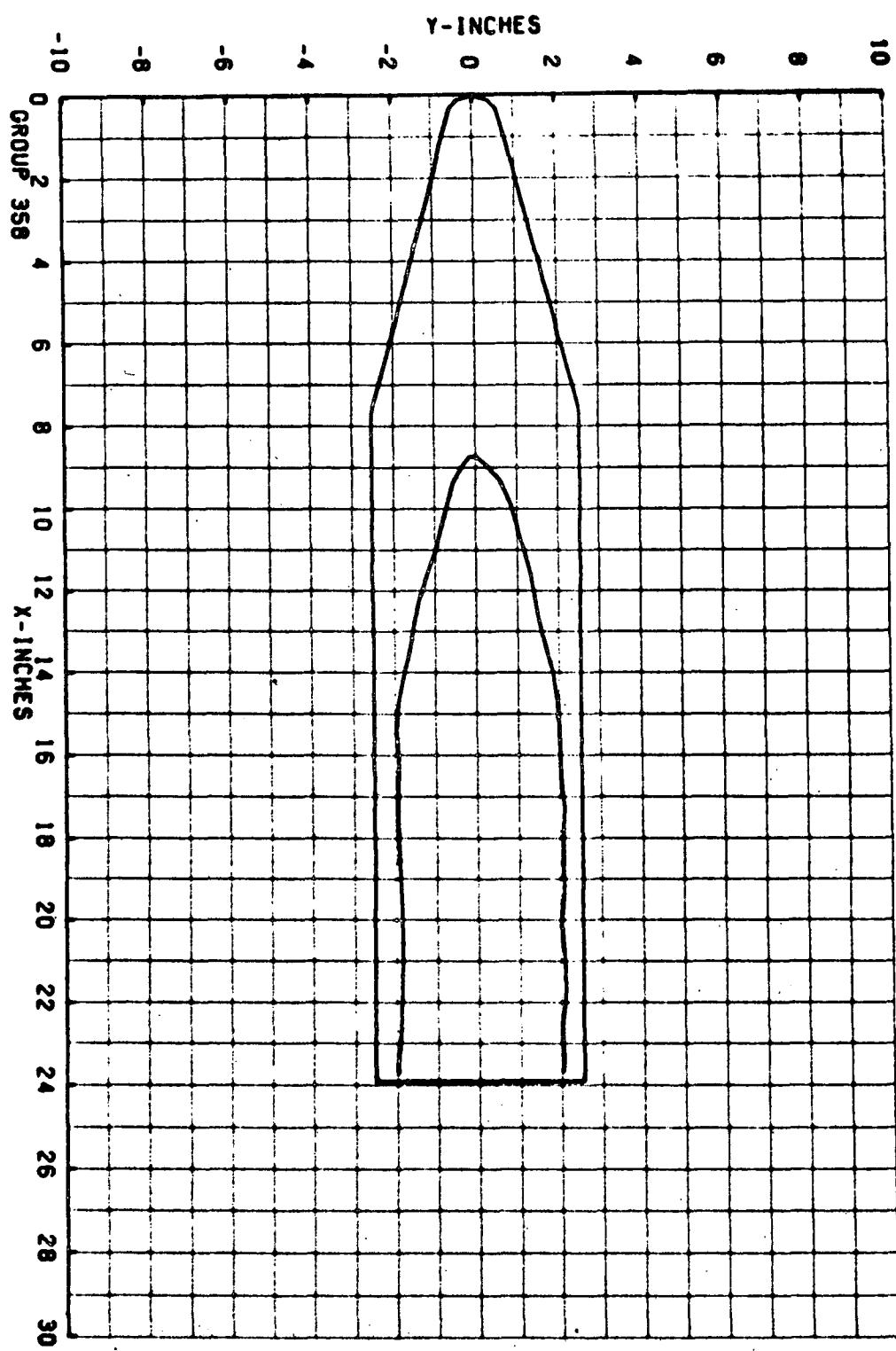
GROUP 143 PIC. NO. 2436 H/HREF 5.650E-02 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 20.0 HREF 2.211E-02 RE/F1 2.520E 06 CONF LRC-58



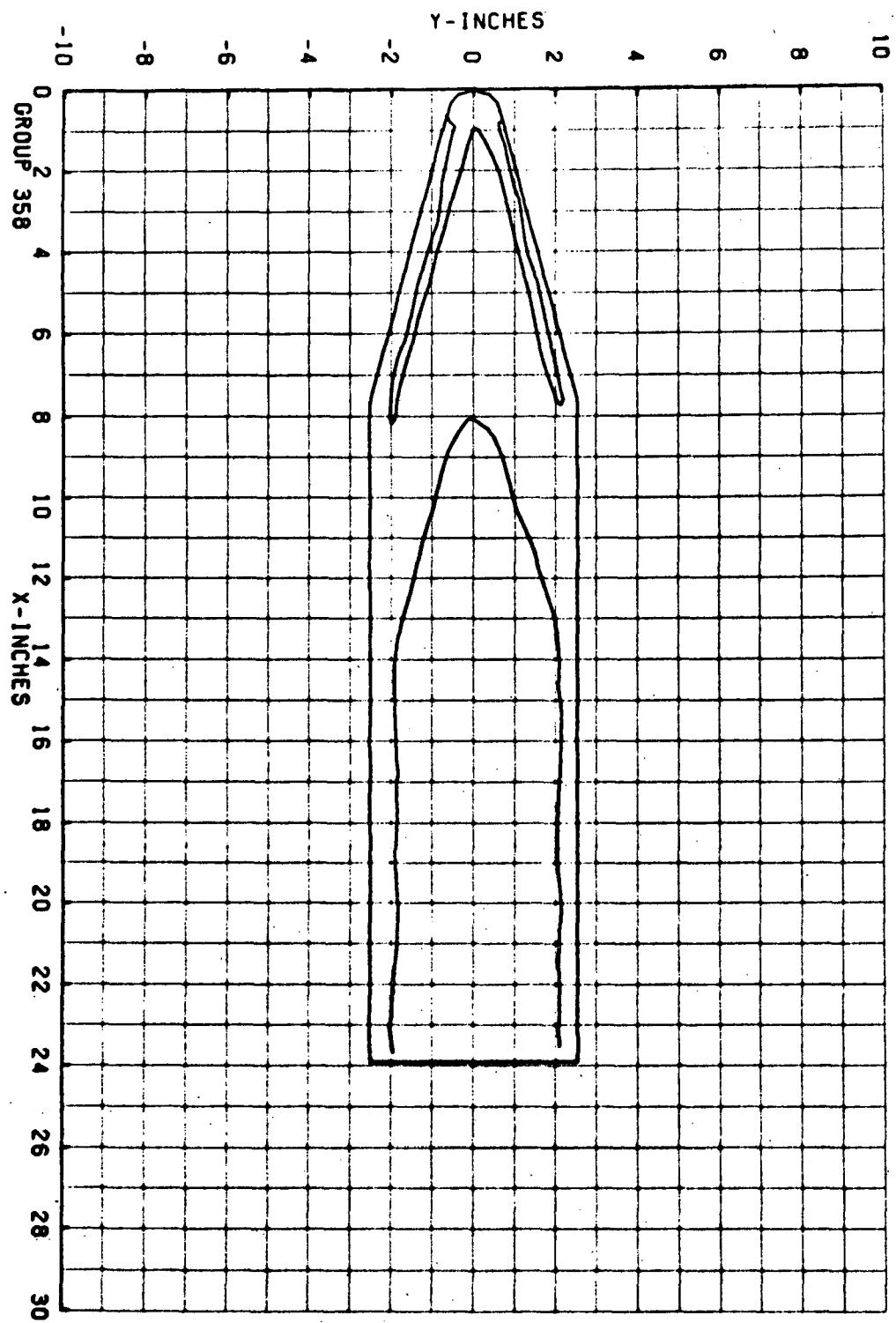
GROUP 143 ALPHA (DEG) 20.0 HREF 2.211E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FT 2.520E 06 CONF LRC-SB



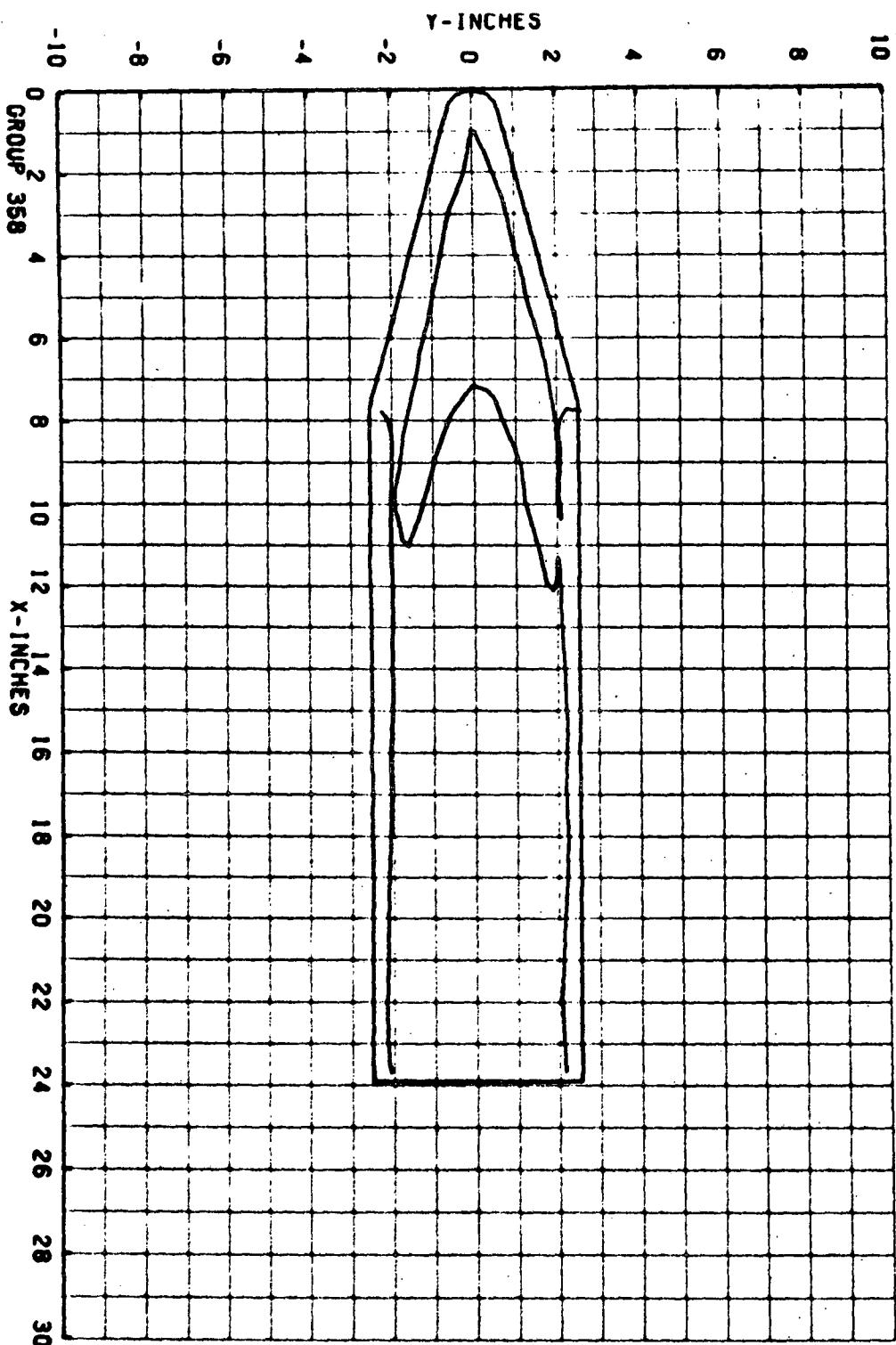
AEDC/AKU, INC., ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL
VILLE



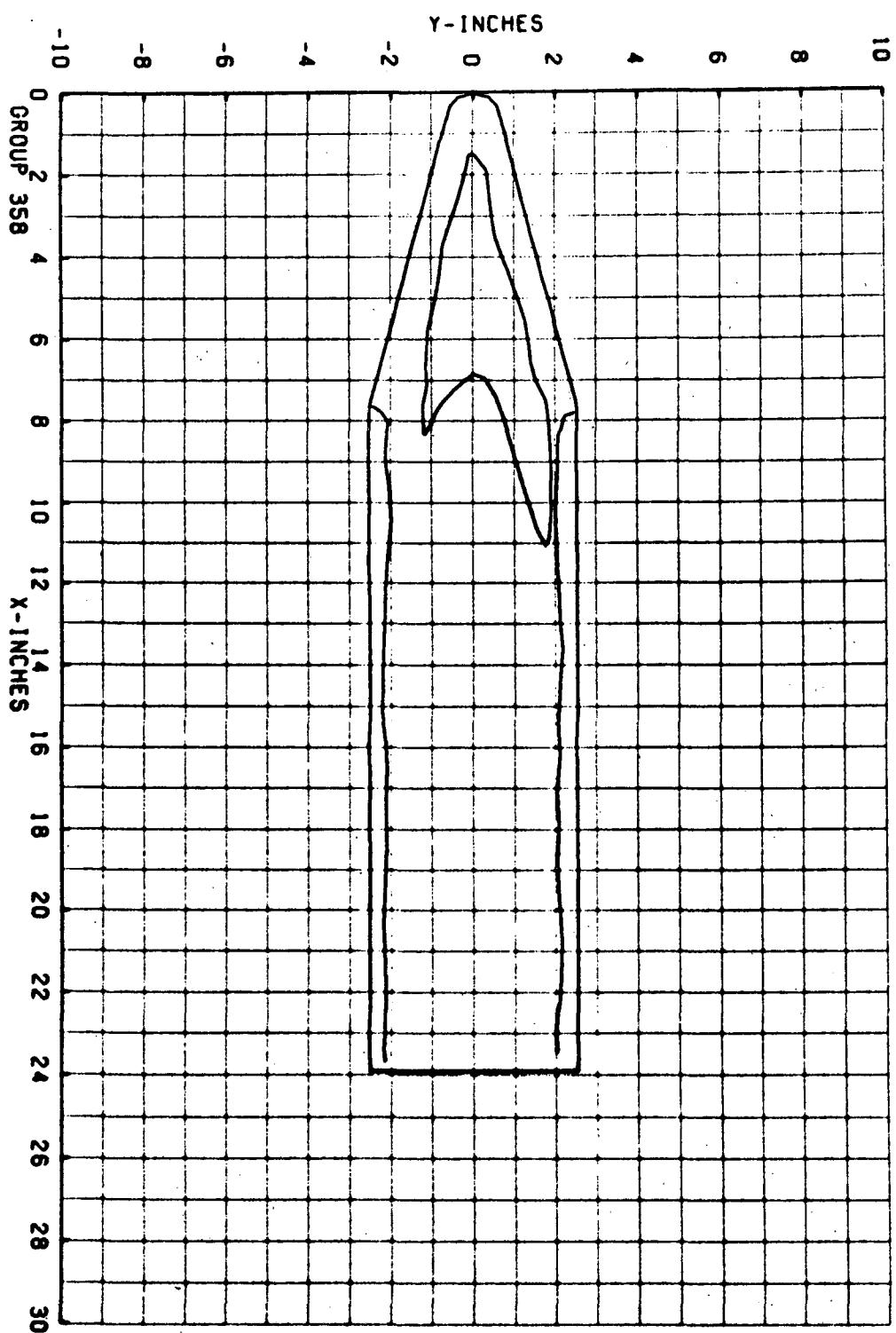
GROUP 358 PIC. NO. 1163 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/HREF 5.704E-01
ALPHA (DEG) 40.0 HREF 2.213E-02 RE/FT 2.520E 06
H/HREF 5.704E-01 HREF 2.213E-02 CONF LRC-SB



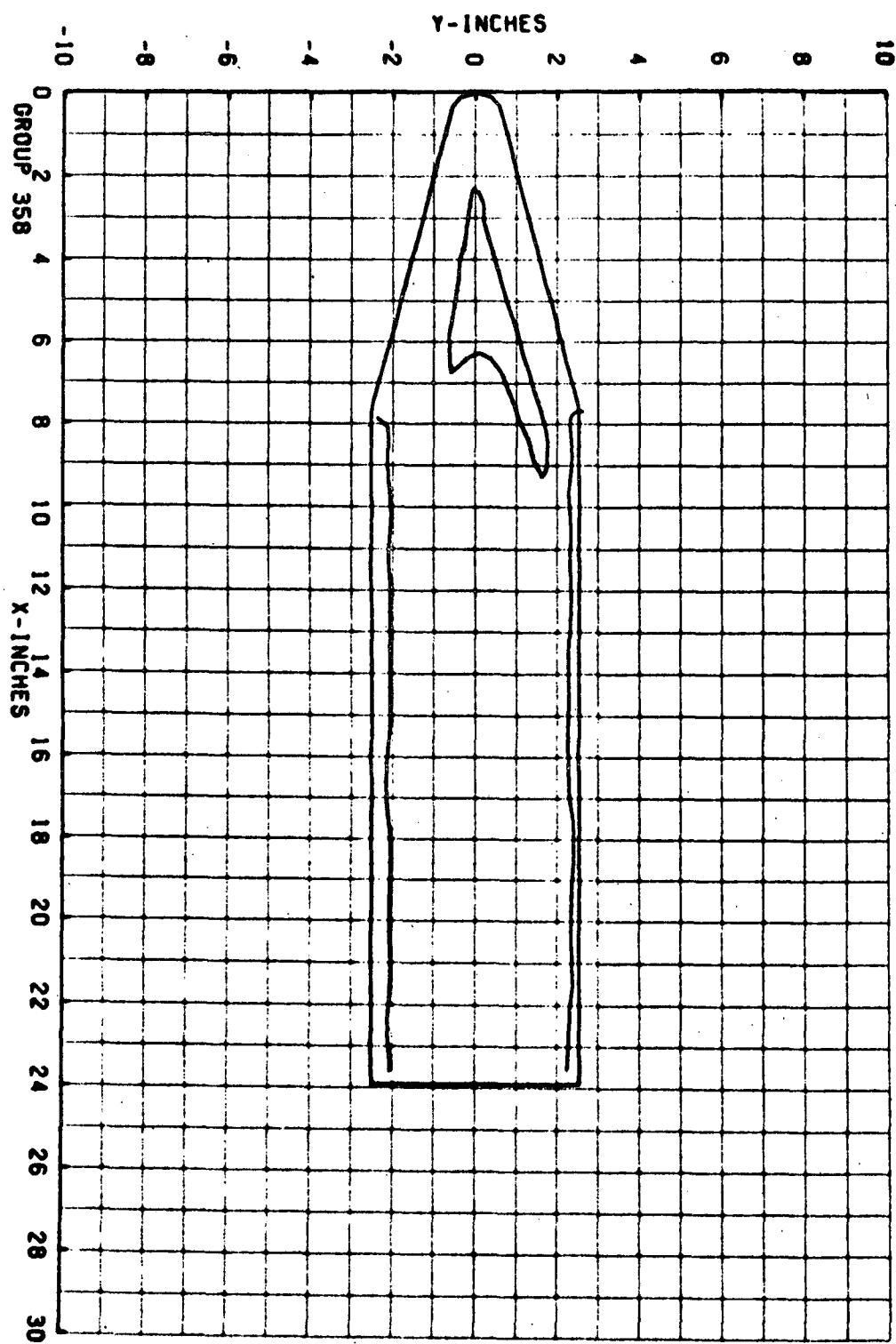
GROUP 358 PIC. NO. 1168 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/HREF 3.974E-01
HREF 2.213E-02 RE/FT 2.520E 06 CONF LRC-SB



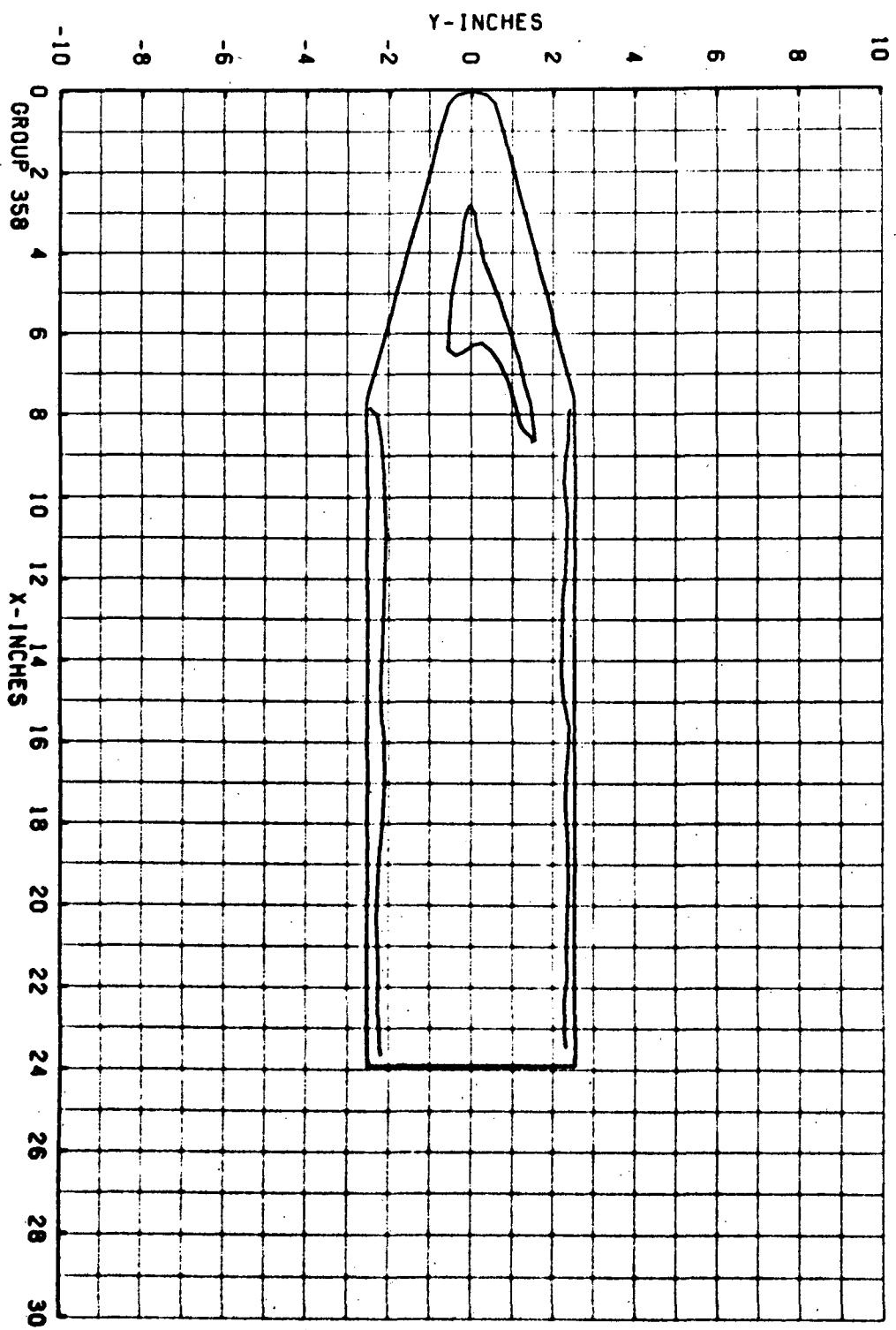
GROUP 358 PIC. NO. 1174 M/HREF 3.021E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.213E-02
GROUP 2 358 RE/FT 2.520E 06 CONF LRC-SB



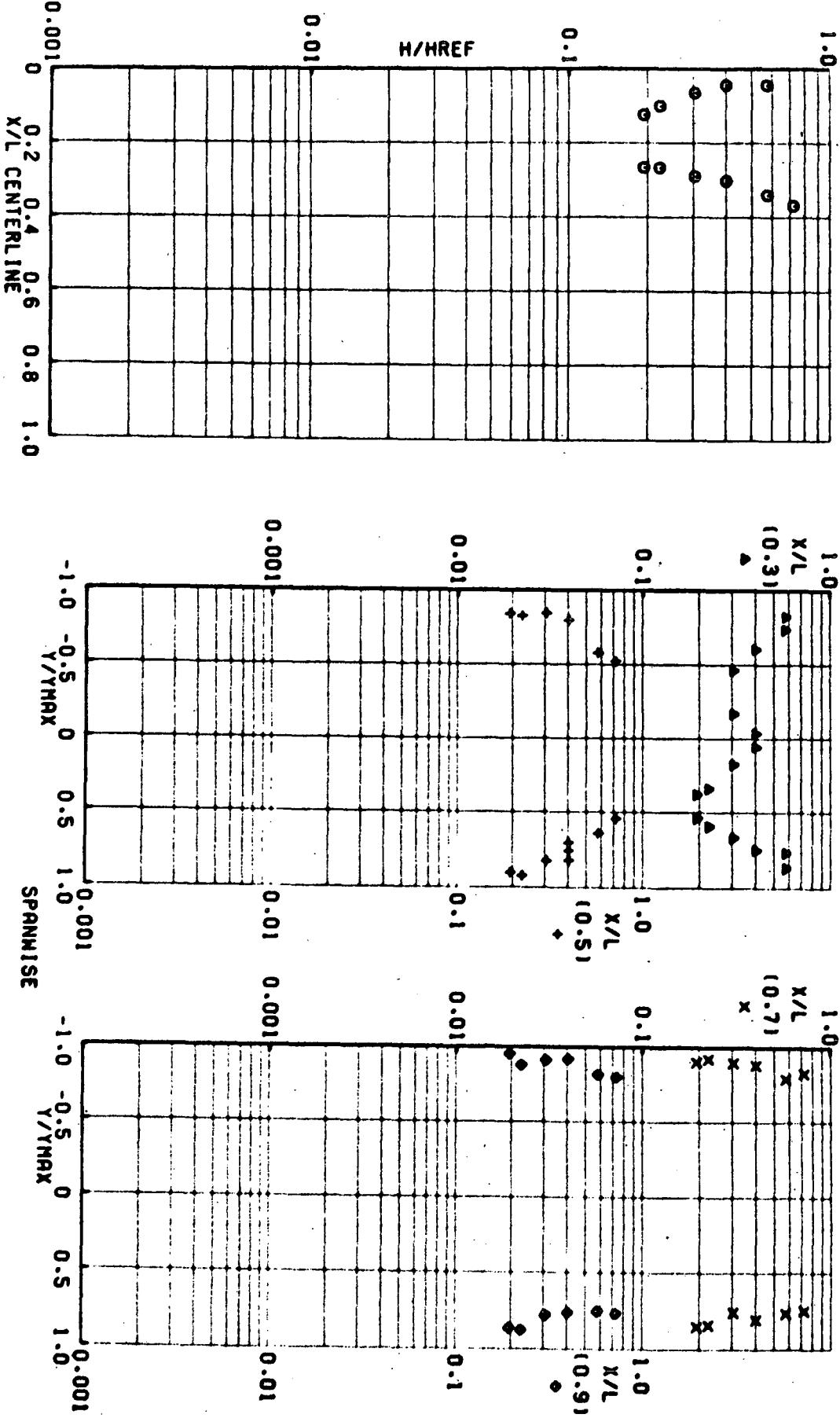
GROUP 358 PIC. NO. 1184 H/HREF 2.217E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.213E-02
MODEL SURFACE - BOTTOM
RE/FT 2.520E 06 CONF LRC-SB



GROUP 358 PIC. NO. 1188 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/HREF 1.928E-01
HREF 2.213E-02 RE/FT 2.520E 06 CONF LRC-58



GROUP 358 ALPHA (DEG) 40.0 HREF 2.213E-02
 MODEL SURFACE - BOTTOM RE/FT 2.520E 06 CONF LRC-SB
 MACH 8.00



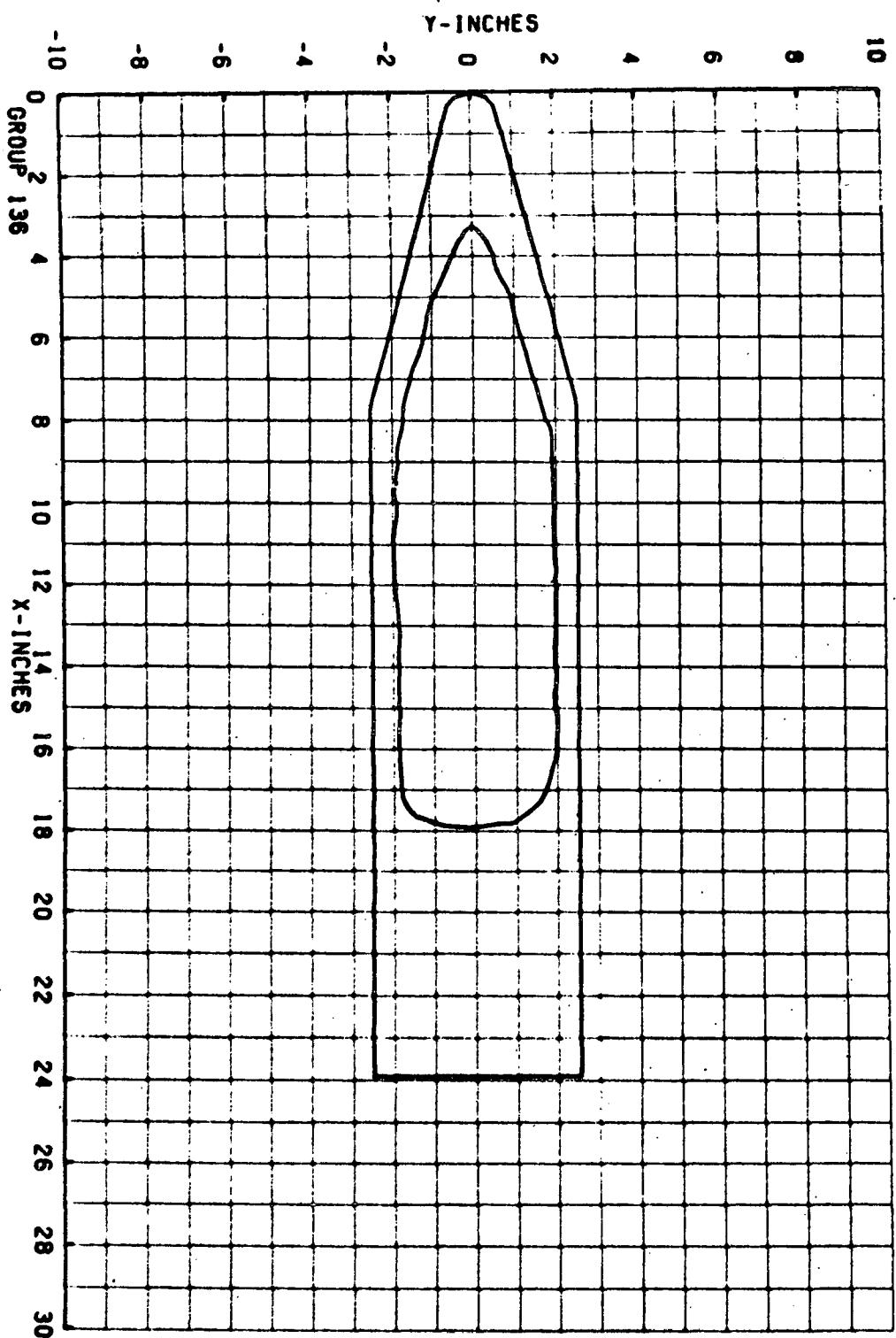
671

**AEDC (AFW, INC.) ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50-INCH HYPERSONIC TUNNEL A**

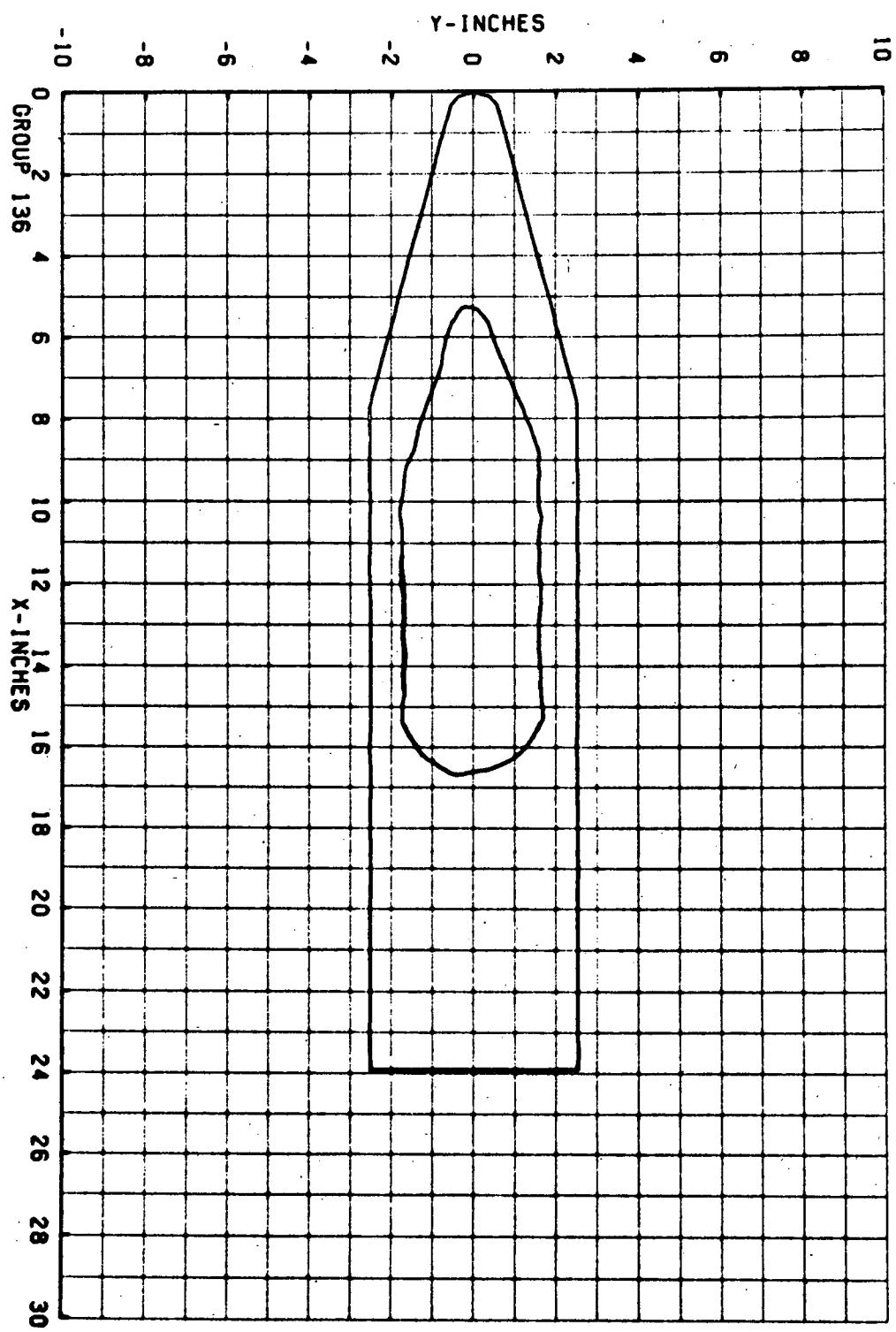
GROUP CONFIG MODEL MACH NO PROPSIA TO DEG R ALPHA-PROTEL ALPHA-SECTOR ALPHA-PREPEN ROLL-MODEL YAW
 136 12 LRC-SM 8.00 553.9 1308 59.98 -9.98 -50.00 180.00 .0 .0
 1-INF P-INF Q-INF V-INF NHO-INF MU-INF RE/FI MHREF SIEF
 (DEG R) (PSI) (PSI) (FT/SEC) (SLUS/FT³) (LB-SEC/FT²) (FT) (R_a=0.056FT) (R_b=0.056FT)
 90.8 .057 2.542 .3H16 5.03E-05 7.63E-08 2.51E-06 2.22E-02 1.436E-02
 CAMERA PAINT TEMP (DEG F) INITIAL TEMP (DEG F) SQUARE HOUT (RHOXCK)
 TOP(1) 200 AVERAGE T_w = 84 = 0.008(SQUARE.HOUT DEL TIME) + 0.11
 SIDE(5) 200
 BOTTOM(8) 200

PIC NO TIME DELAYE HROUT HROUT/HREF HROUT/HREF HROUT/HREF SIEF
 1 2.78 (200) 3.70 2.66 9.01E-03 .4121 1.132F-02 .5122 1.289E-02 .5831 5.944E-03 0 84 0 0
 1 2180 (200) 4.80 3.76 7.04E-03 .3378 9.20U-03 .4199 1.057E-02 .4780 4.840E-03 0 84 0 0
 1 2181 (200) 5.30 4.26 6.94E-03 .3140 8.624E-03 .3903 9.822E-03 .4444 4.500E-03 0 84 0 0
 1 2182 (200) 5.85 4.81 6.06E-03 .2923 8.030E-03 .3633 9.422E-03 .4136 4.180E-03 0 85 0 0
 1 2183 (200) 6.40 5.36 6.05E-03 .2740 7.521E-03 .3405 8.569E-03 .3877 3.926E-03 0 85 0 0

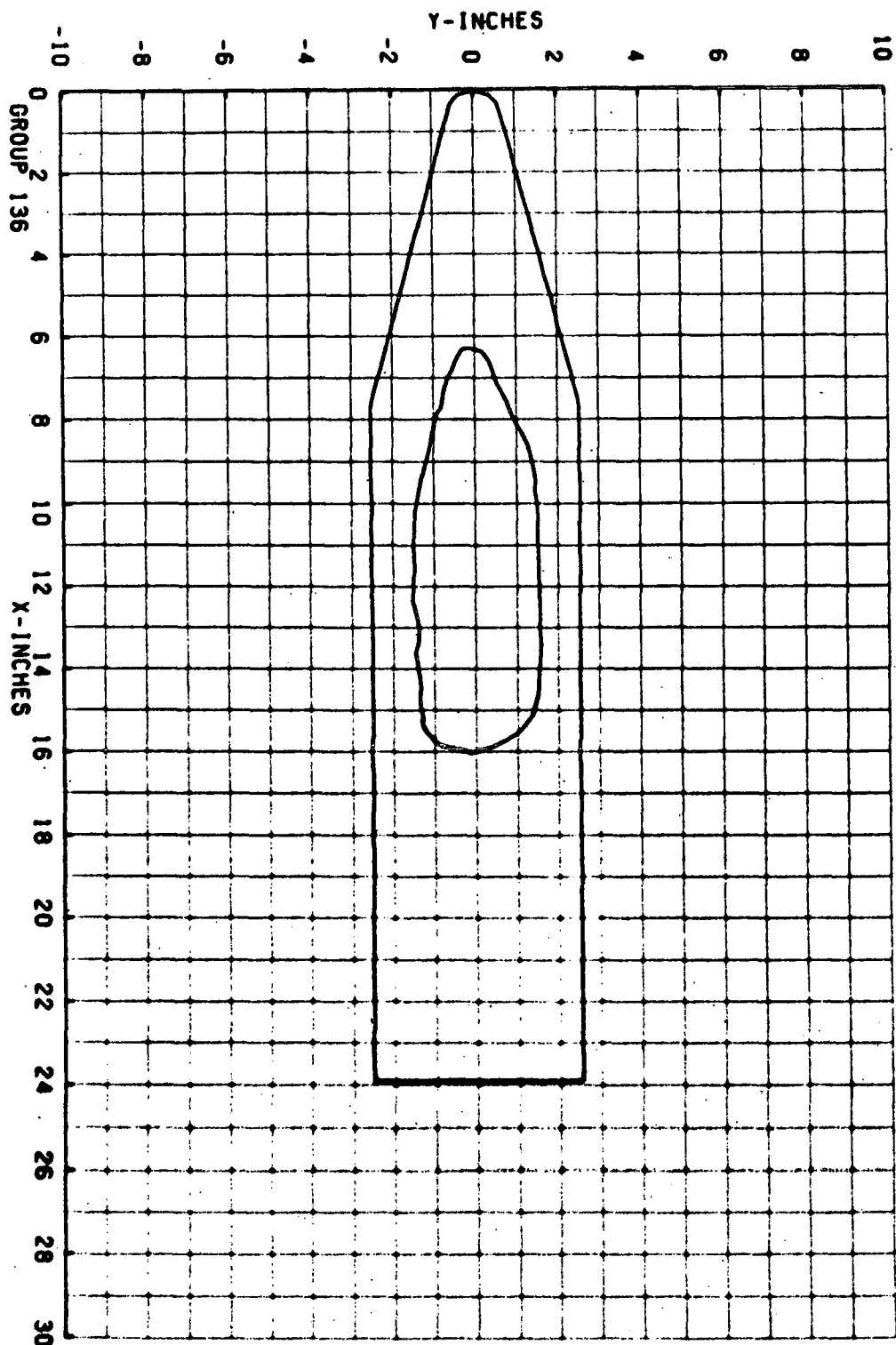
GROUP 136 PIC. NO. 2178 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 60.0 H/HREF 4.121E-01
ALPHA (DEG) 60.0 HREF 2.210E-02 RE/FT 2.510E 06 CONF LRC-SB



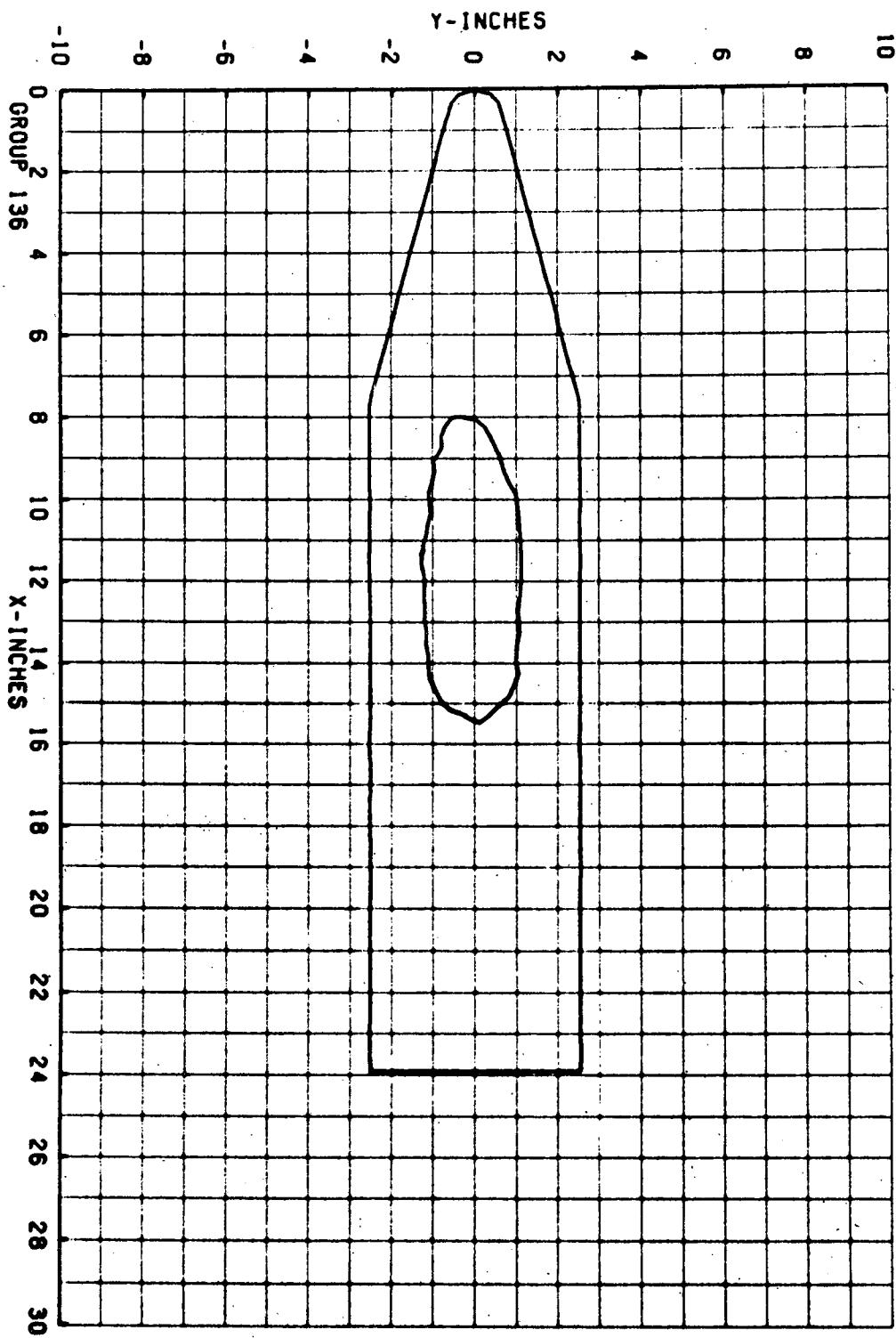
GROUP 136 PIC. NO. 2180 H/HREF 3.378E-01
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.210E-02
RE/FI 2.510E 06 CONF LRC-SB



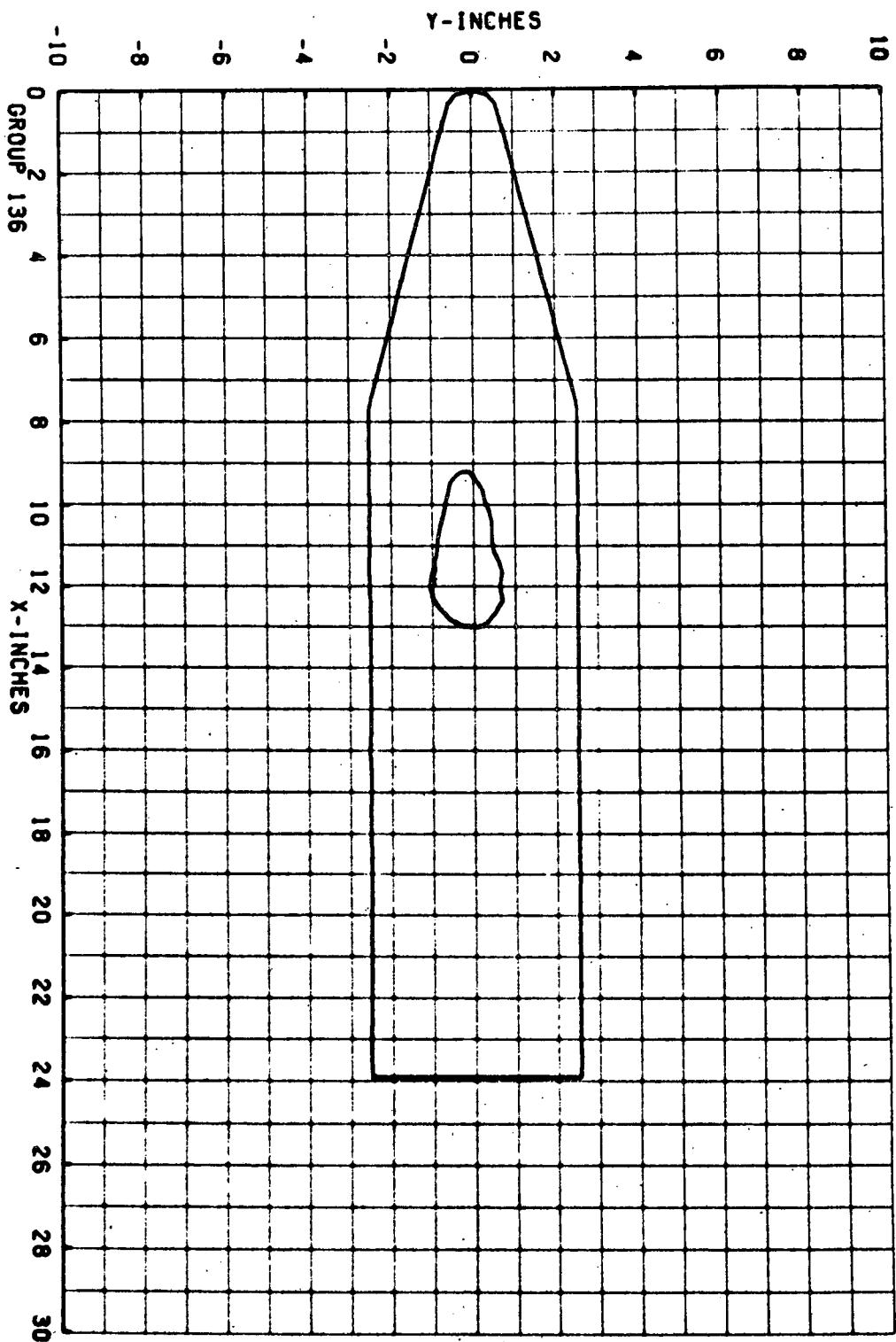
GROUP 136 PIC. NO. 2181 H/HREF 3.140E-01
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.210E-02
ALPHA (DEG) 60.0 HREF 2.210E-02
RE/F1 2.510E 06 CONF LRC-SB



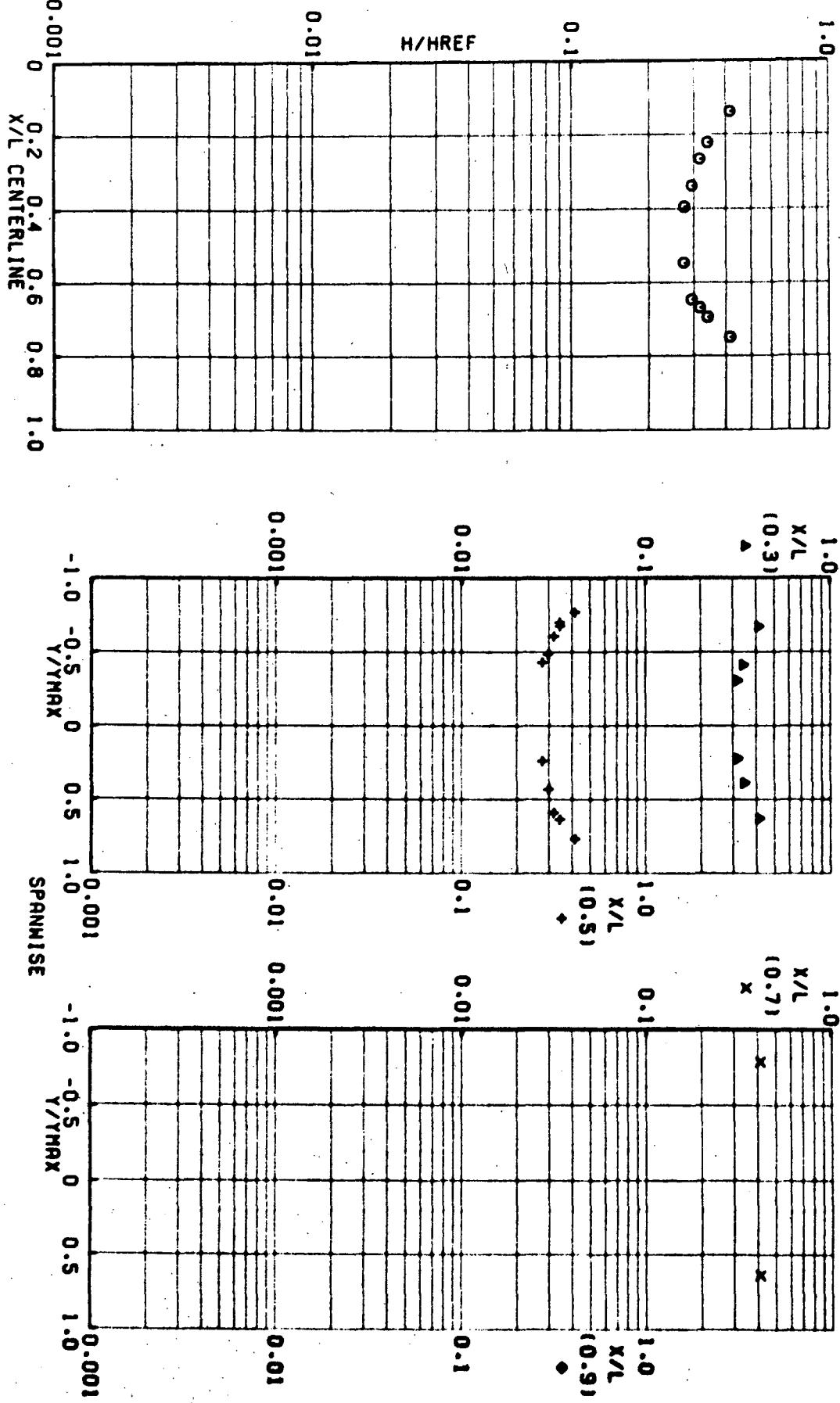
GROUP 136 PIC. NO. 2162 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 60.0 H/HREF 2.923E-01
HREF 2.210E-02 RE/FT 2.510E-06 CONF LRC-50



GROUP 136 PIC. NO. 2183 H/HREF 2.740E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.210E-02 RE/F1 2.510E 06 CONF LRC-SB



GROUP 136 ALPHA (DEG) 60.0 HREF 2.210E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FT 2.510E 06 CONF LRC-SB

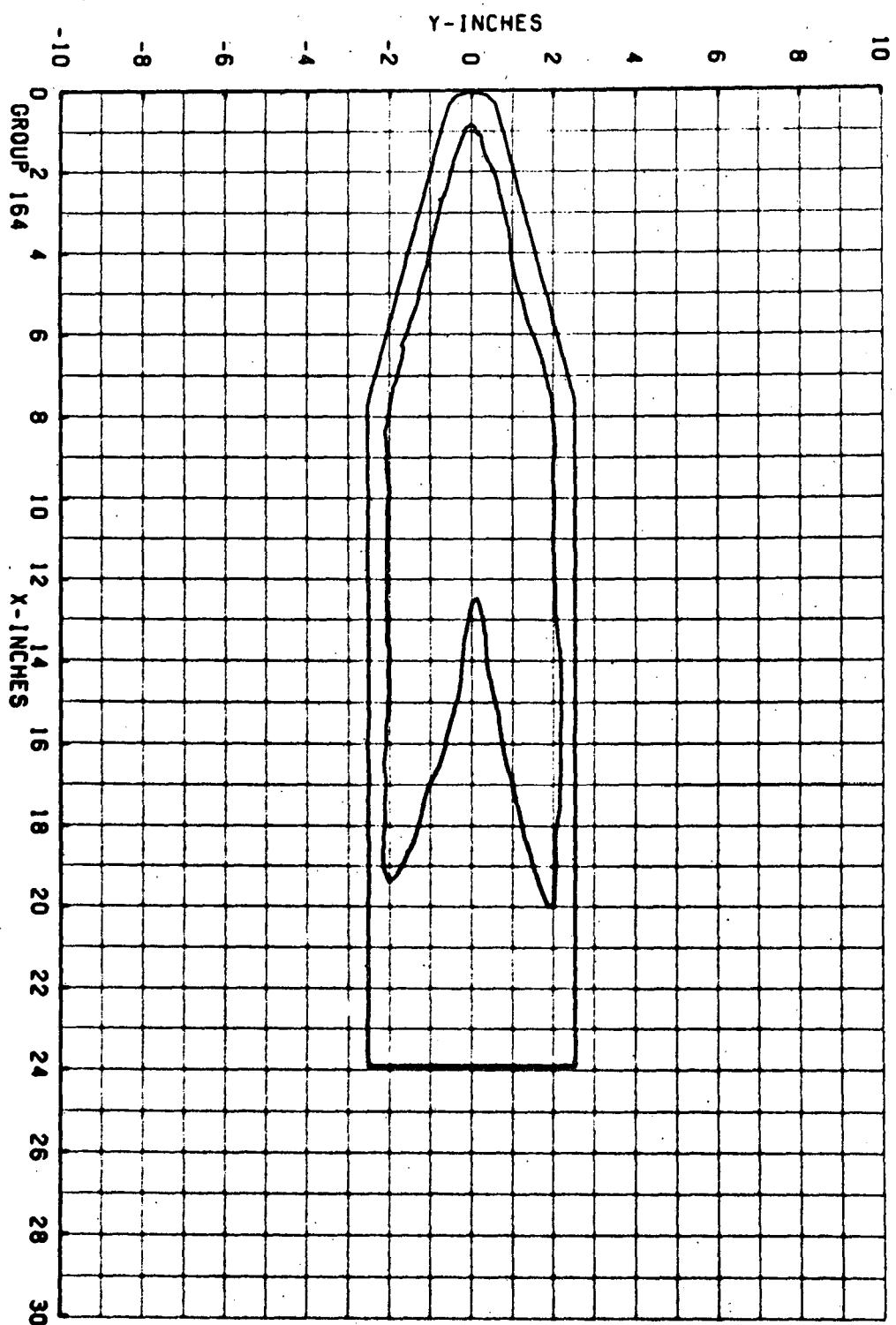


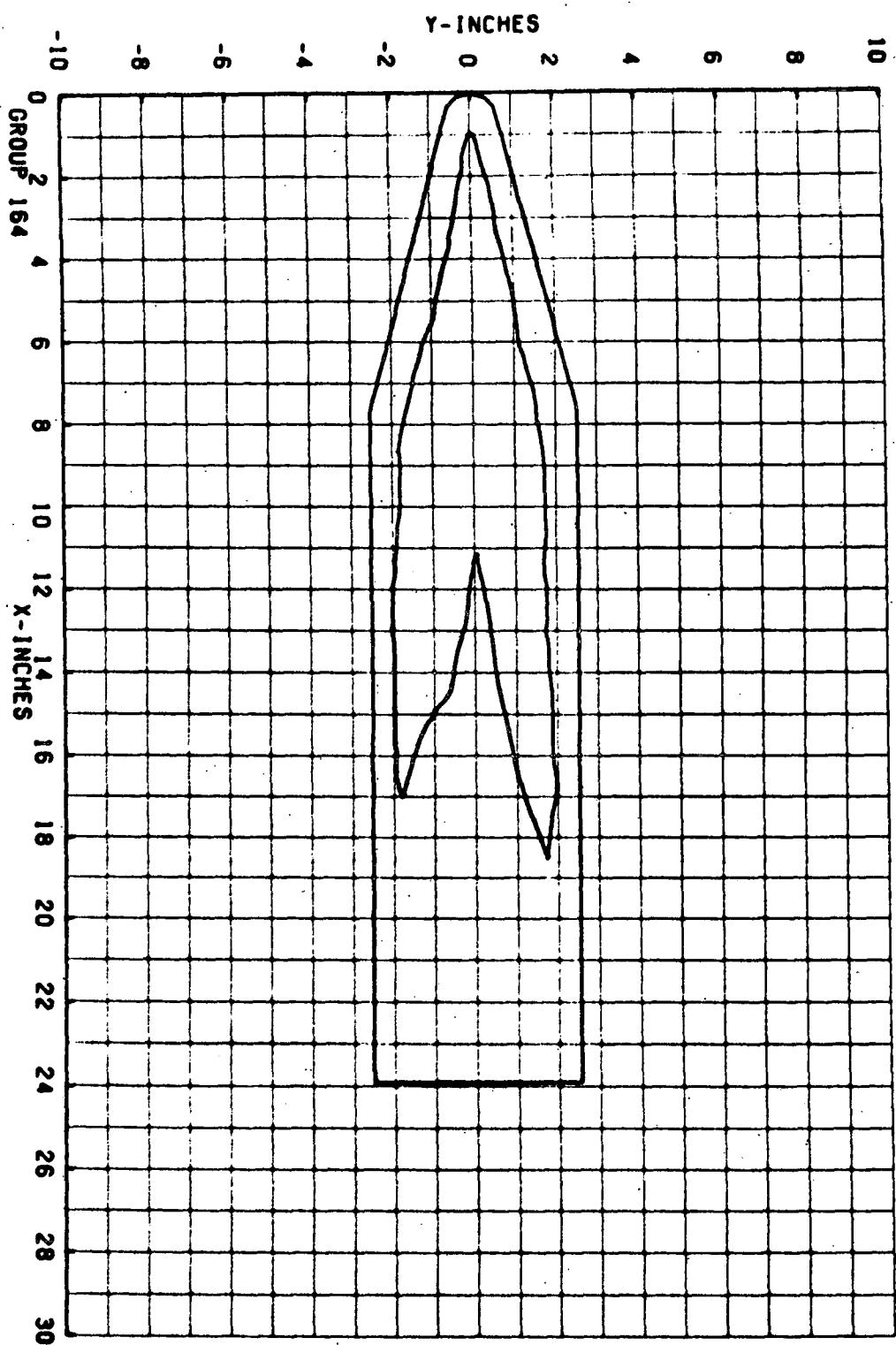
1371

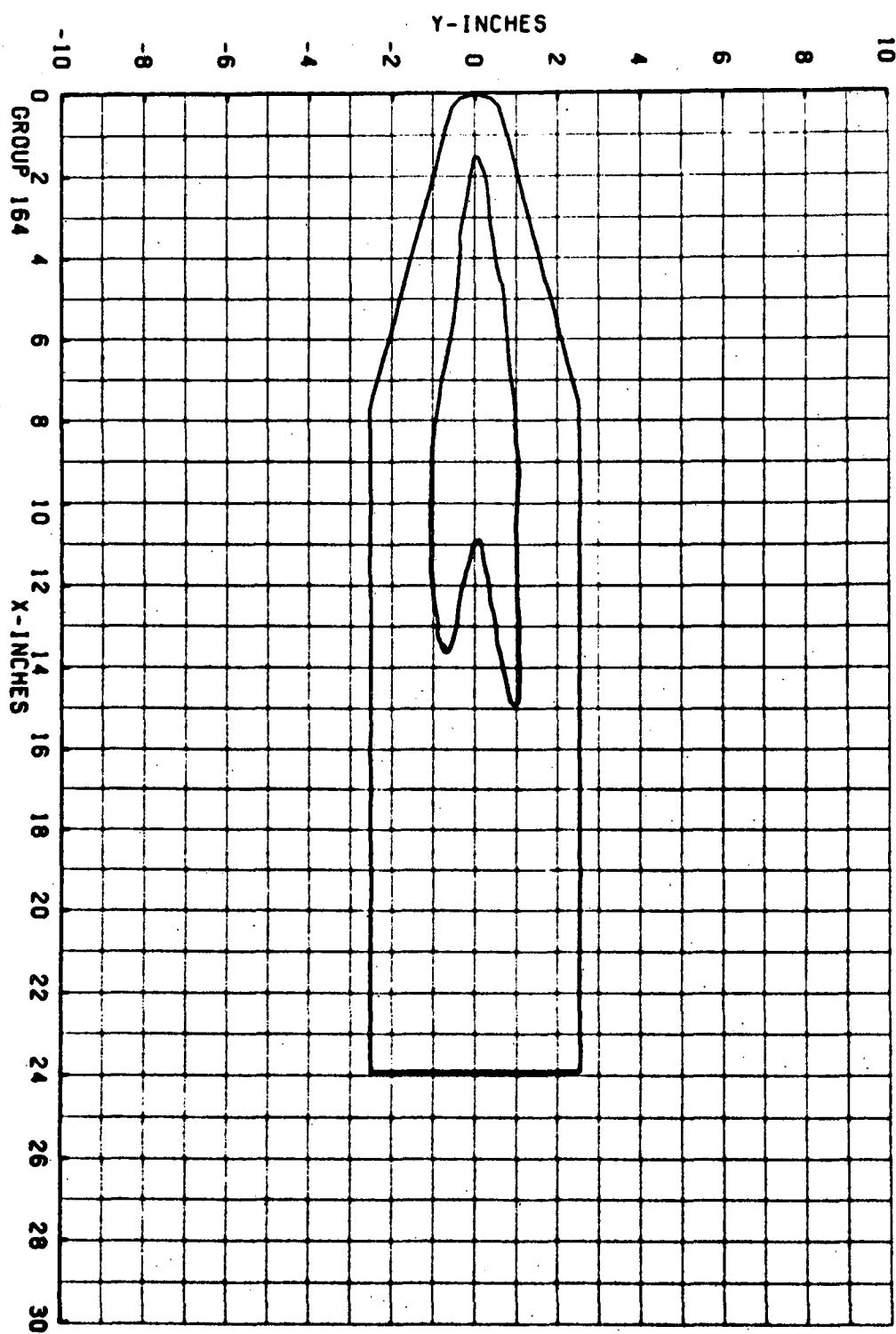
AEDC (ARO, INC.) ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL B

GROUP 10010 MODEL MACH NO 0.00 P0/PSIX 10 DEG R ALPHA=0.00 DEL ALPHA=SECTOR ALPHA=PREBEND ROLL-MODEL VAW
 164 12 LRC-SB 8.00 856.4 1351 20.00 3.00 -23.00 180.00 .0
 T-INF P-INF Q-INF V-INF RHO-INF MU-INF RE/FT HREF STREF
 (NEG R) (PSIA) (PSIA) (FT/SEC) (SLUGS/FT³) (LB-SEC/FT²) (FT-1) (R_a = 0.056FT) (R_b = 0.056FT)
 97.9 .008 3.920 3878 1.518E-02 7.891E-08 3.10E-06 2.763E-02 1.171E-02
 CAMERA PAINT TEMP (DEG F) INITIAL TEMP (DEG F) SQUARE ROOT (RMOKCXK)
 TOP(1) 163 163 AVERAGE TEMP = 70 = 0.0081 SQUARE ROOT DEL TIME & DEL
 SIDE(S) BOTTOM(B) 163

GROUP 164 PIC. NO. 3322 H/HREF 2.394E-01
MACH 8.00 ALPHA (DEG) 20.0 HREF 2.763E-02
RE/FT 3.700E 06 CONF LRC-SB

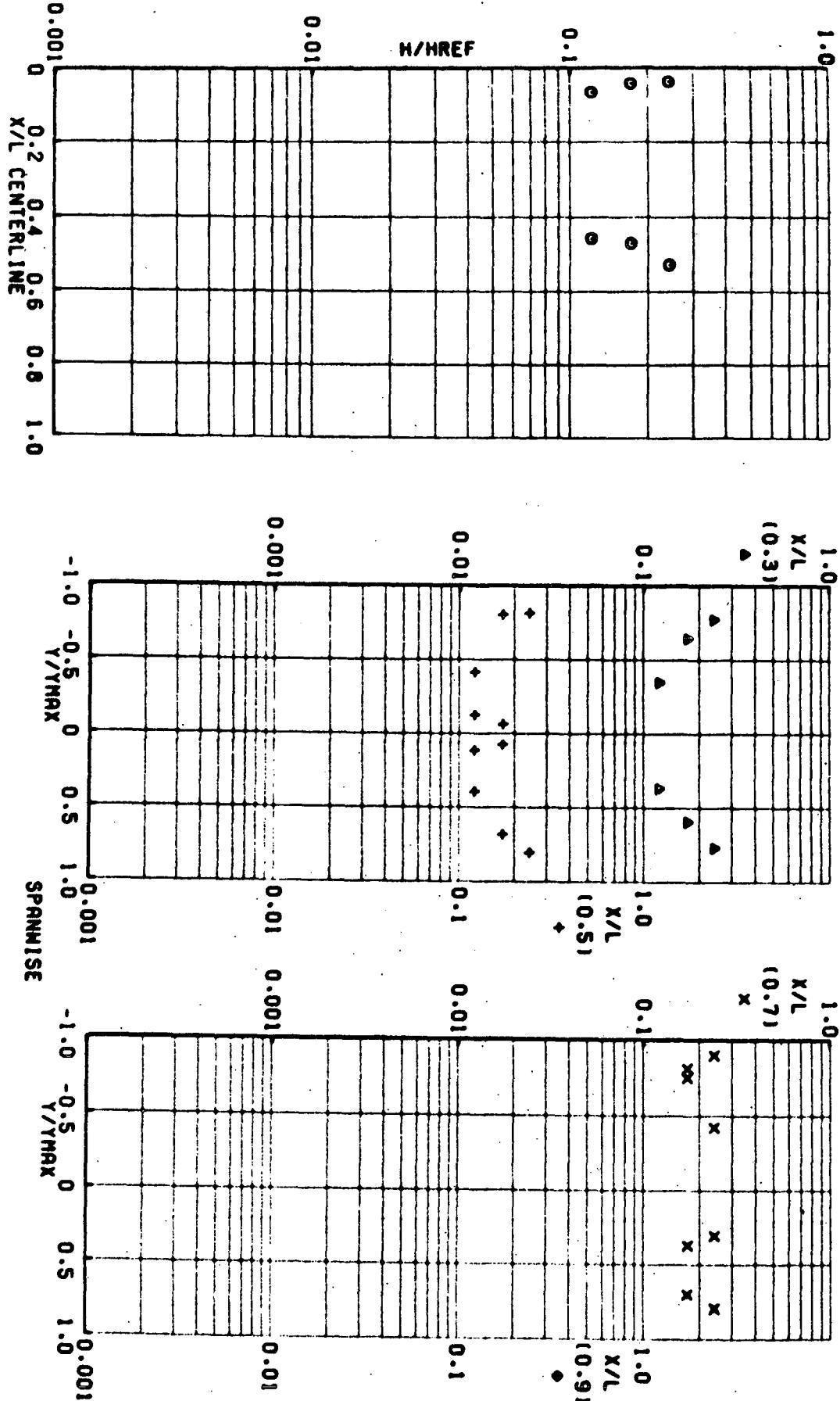






GROUP 164 PIC. NO. 3333 H/HREF 1.203E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 20.0 HREF 2.763E-02 RE/FT 3.700E 06 CONF LRC-5B
ALPHA (DEG) 20.0 HREF 2.763E-02 RE/FT 3.700E 06 CONF LRC-5B

GROUP 164 ALPHA (DEG) 20.0 HREF 2.763E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FT 3.700E 06 CONF LRC-SB

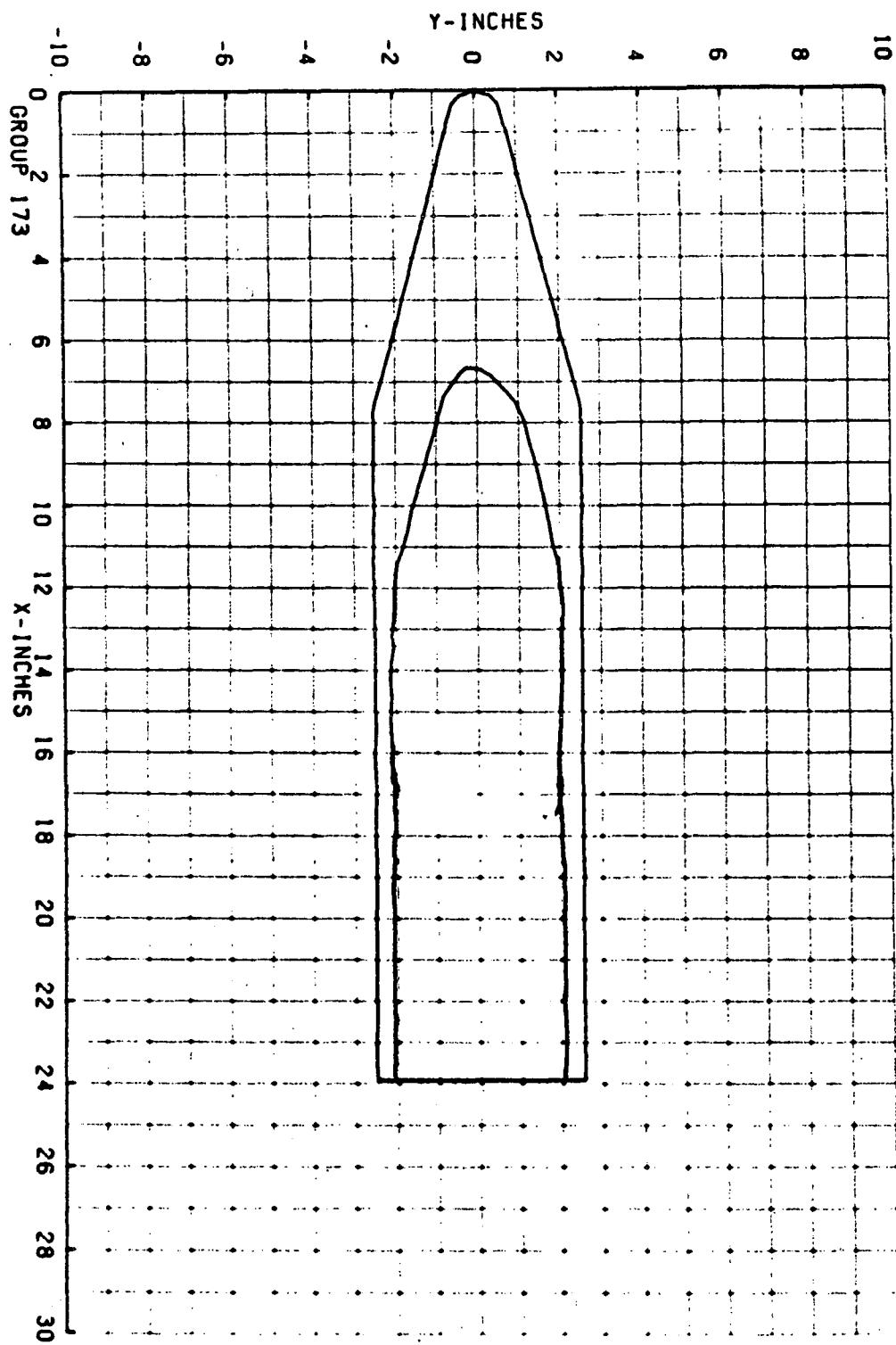


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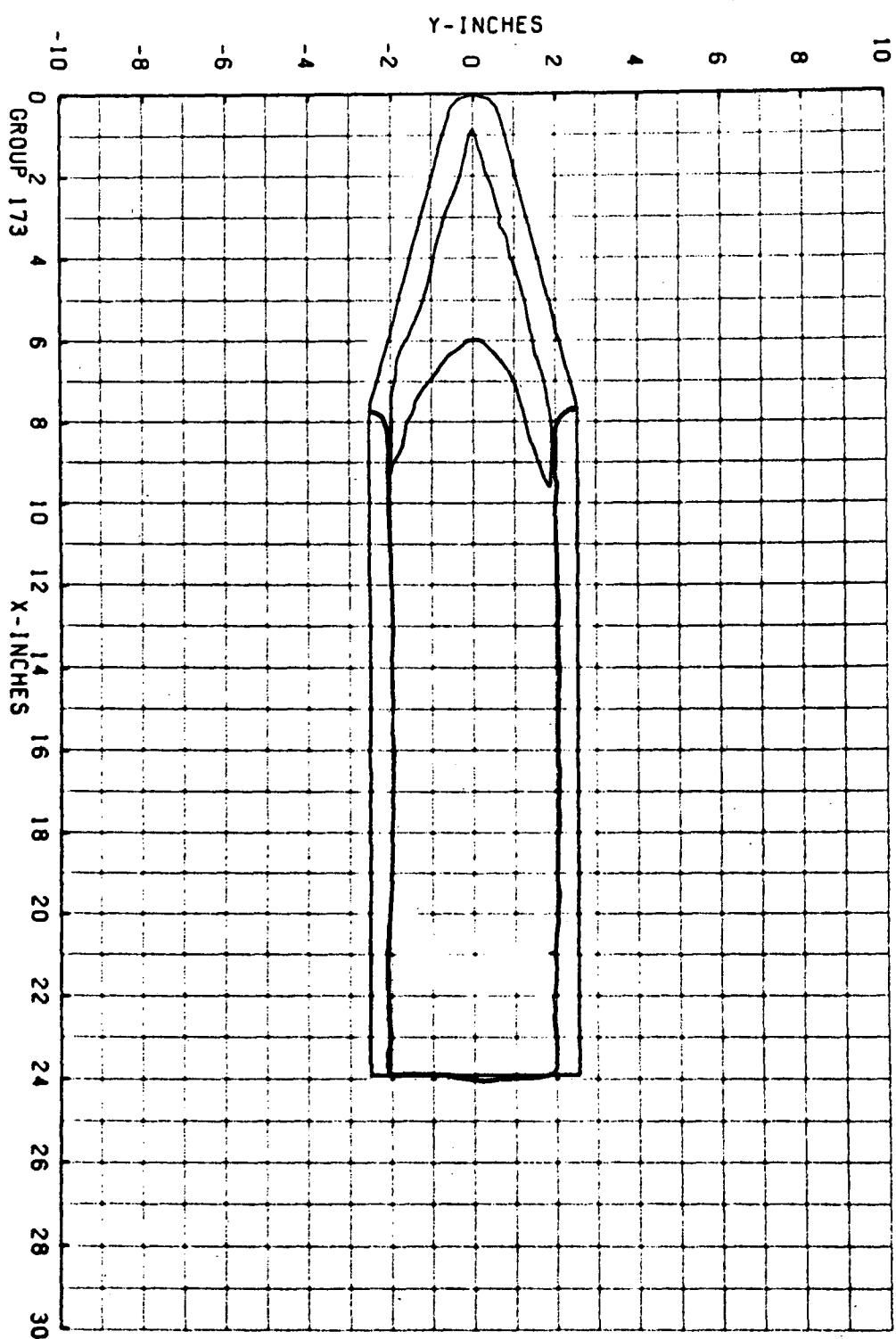
AFDC (ARO INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL B
 VII162

GROUP	CONFIG	MODEL	MACH NO	P0	PSTA	T0	DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAN
173	12	URC-5A	8.00	858.6		1346	39.99	10.01	-50.00	180.00	0	
		I-INF P-INF	0-INF	V-INF	RHO-INF	MU-INF	REF/FT	HREF	STREF			
		(DEG R)	(PSTA)	(PSTA)	(SLURS/FT3)	(H-SEC/FT2)	(FT-1)	(H = .056FT)	(H = .056FT)			
		97.6	.000	3.940	3P72	7.564E-05	7.854E-08	2.765E-02	1.173E-02			
		CAMERA		PAINT TEMP (DEG F)		INITIAL TEMP (DEG F)		SQUARE ROOT (RHO*CXK)				
		TOP(1)	250			AVERAGE TM = 73		-0.008(SQUARE ROOT (FL TIME) + 0.11				
		SIDE(1)	100									
		BOTTOM(1)	100									
		PIC NC	TIME DELTIME	H(T0)	H(T0)/HREF	H(.9T0)-H(.9T0)/HREF	H(.85T0)	H(.85T0)/HREF	ST(T0)	MODEL TEMP F		
		1 3581	(250) 2.60	1.56	1.97E-02	.6771	2.344E-02	.8488	2.689E-02	.9723	7.908E-03	73 0 0
		1 3585	(250) 4.71	3.66	1.66E-02	.4198	1.452E-02	.5249	1.663E-02	.6613	4.892E-03	74 0 0
		1 3592	(250) 8.35	7.31	7.65E-03	.2765	9.587E-03	.3466	1.098E-02	.3971	3.229E-03	80 0 0
		1 3594	(250) 9.40	8.36	7.03E-03	.2561	8.812E-03	.3196	1.009E-02	.3650	2.968E-03	83 80 0 0

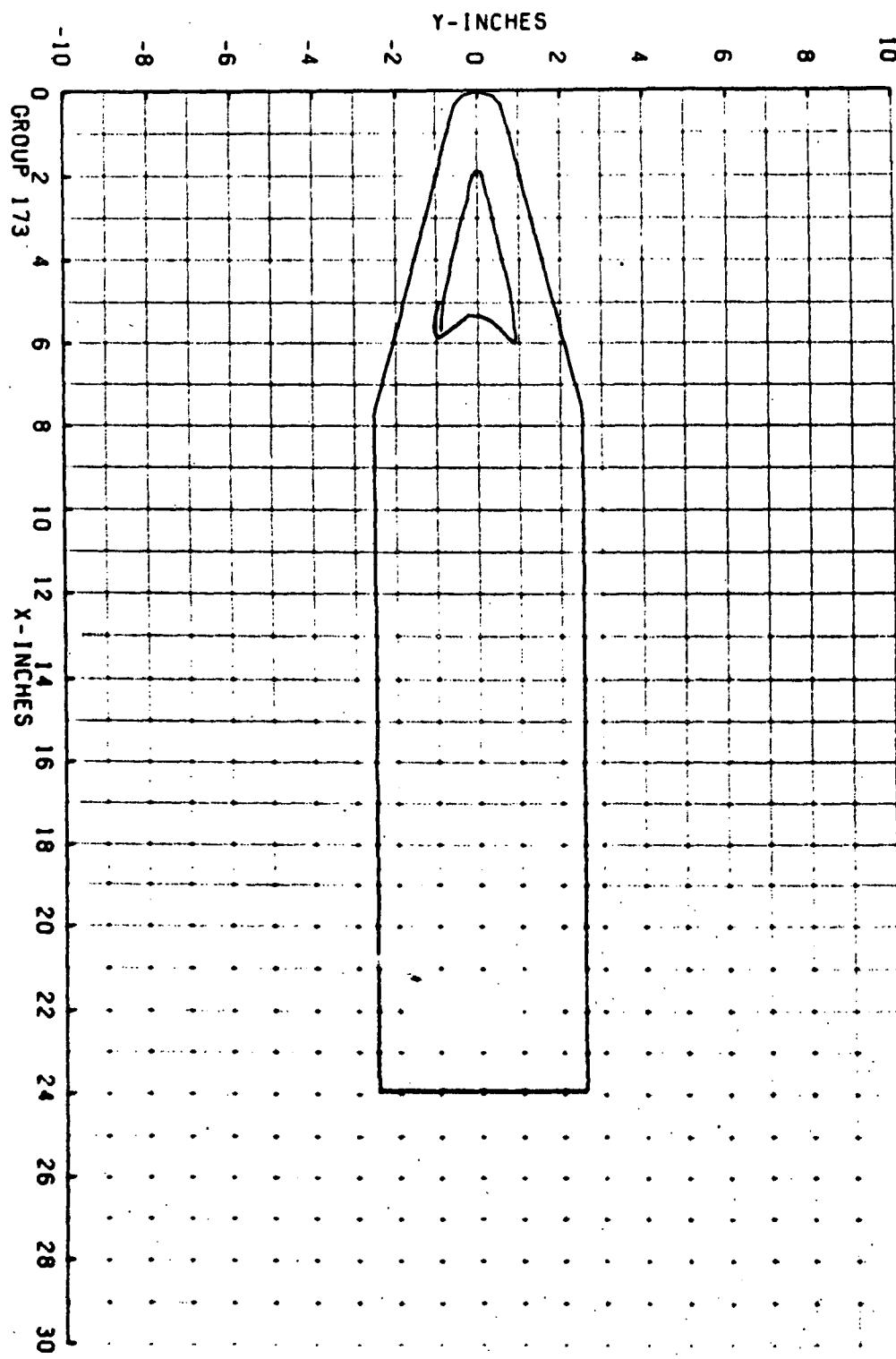
GROUP 173 PIC. NO. 3581 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/HREF 6.771E-01
HREF 2.765E-02 RE/FI 3.730E 06 CONF LRC-S8



GROUP 173 PIC. NO. 3585 H/HREF 4.188E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.765E-02
REFIT 3.730E 06 CONF LRC-SB
MODEL SURFACE - BOTTOM

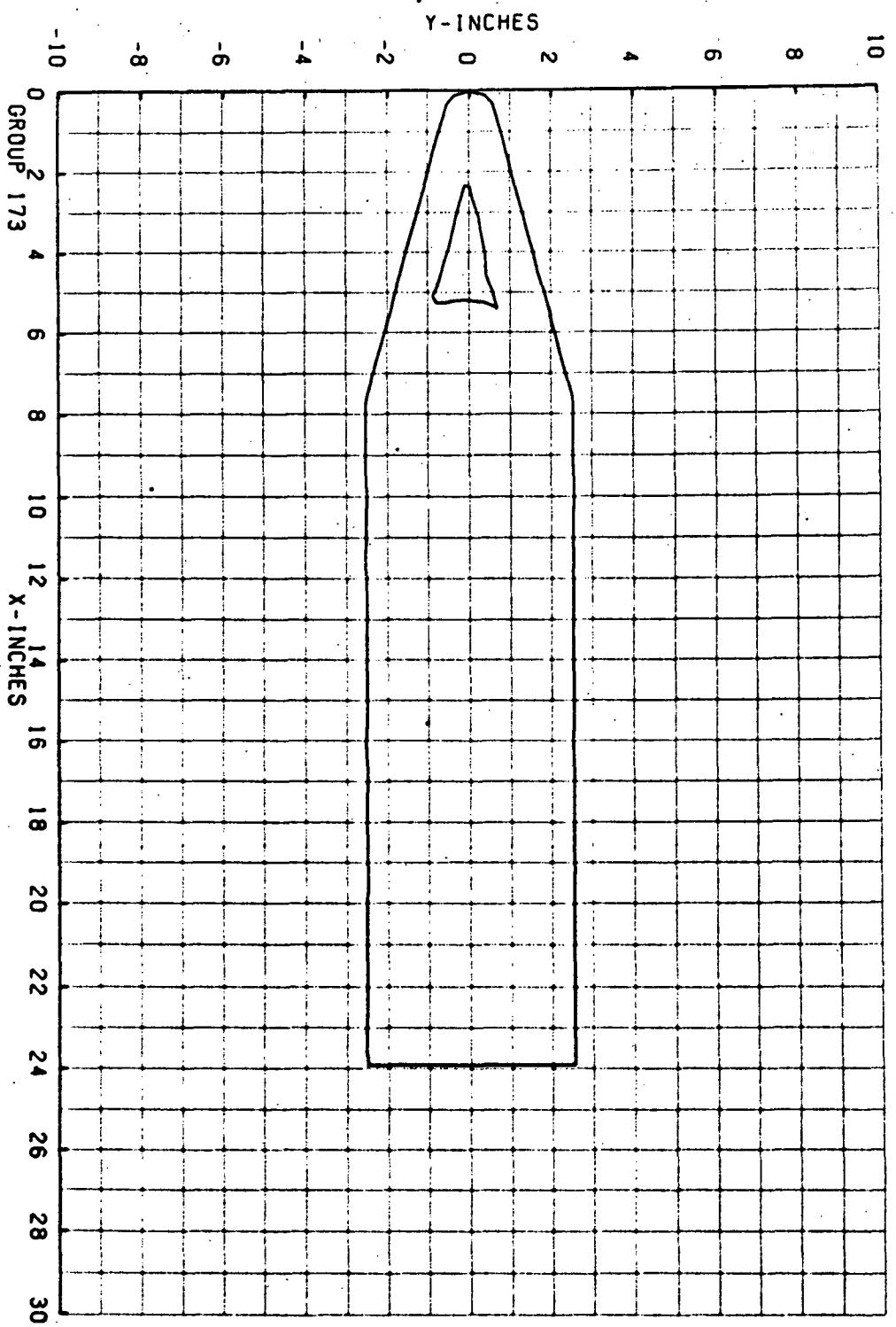


GROUP	173	PIC. NO.	3592	H/HREF	2.765E-01	MODEL SURFACE - BOTTOM
MACH	8.00	ALPHA (DEC)	40.0	HREF	2.765E-02	REFIT 3.730E 06
						CONF LRC-SB

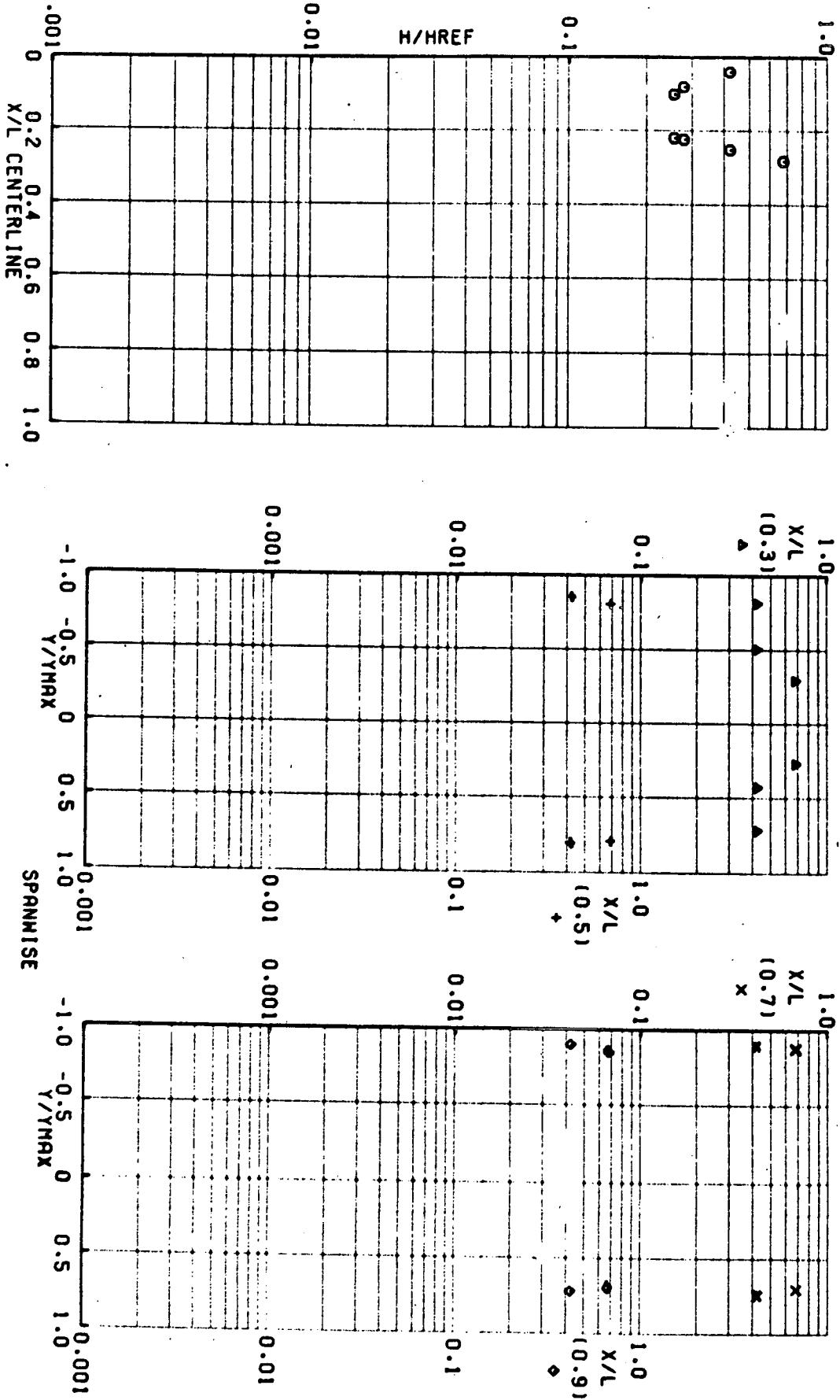


104

GROUP 173 PIC. NO. 3594 H/HREF 2.541E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.765E-02
RE/FT 3.730E 06 CONF LRC-SB



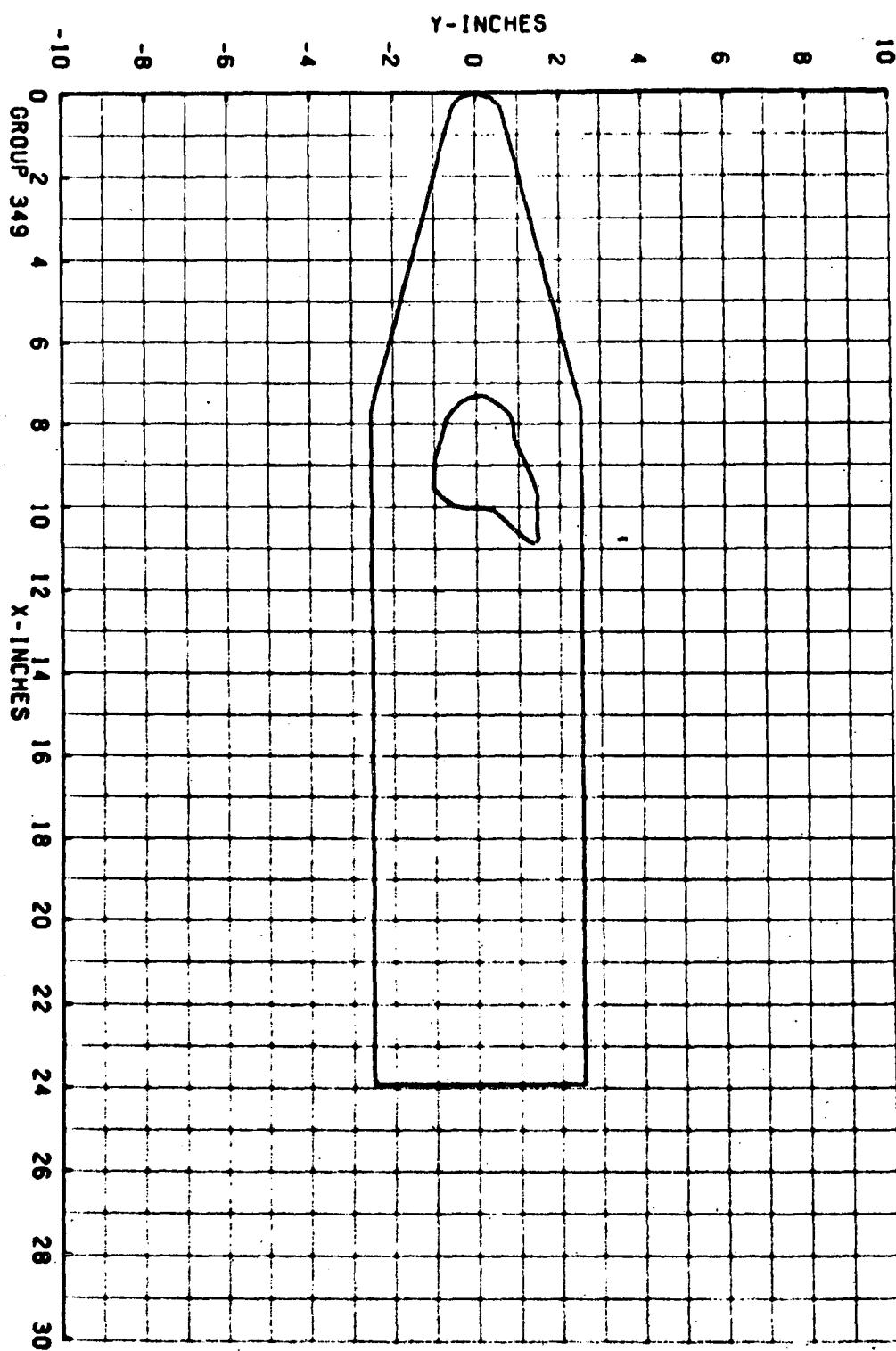
GROUP 173 ALPHA (DEG) 40.0 HREF 2.765E-02 MACH 8.00



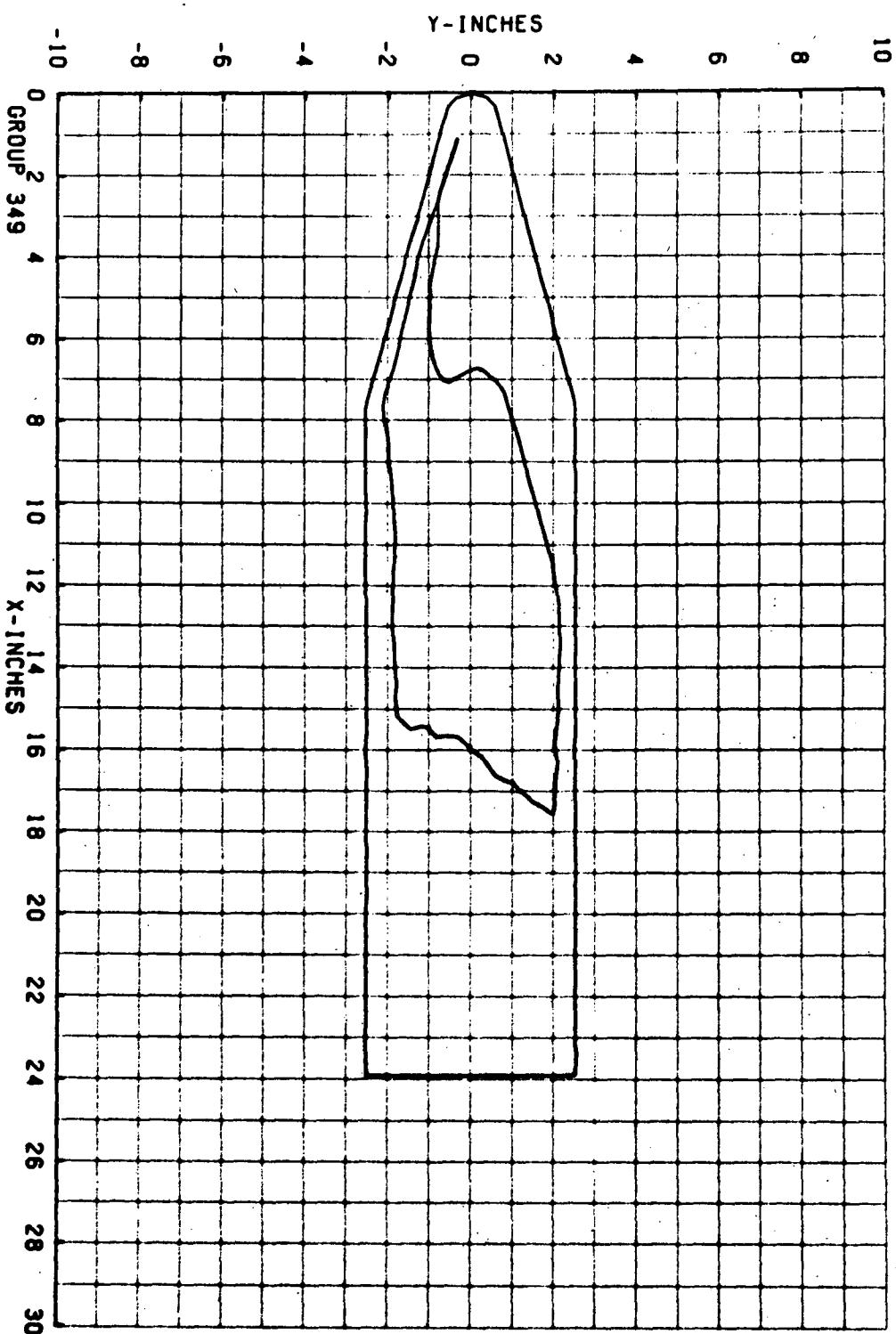
137

AEDC/AMU, INC., ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL B
VT1162

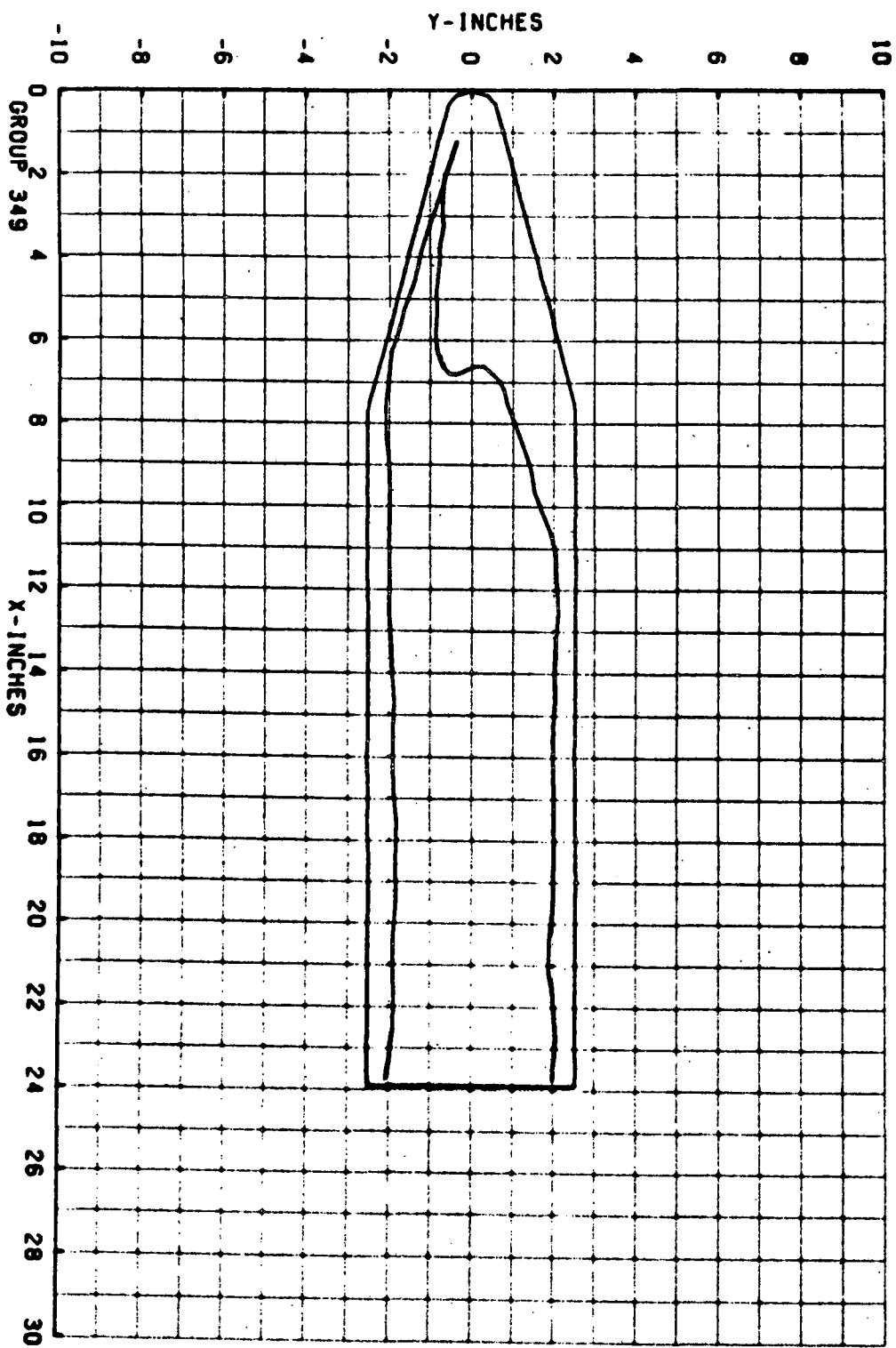
GROUP 349 PIC. NO. 869 H/HREF 7.746E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.768E-02 RE/FT 3.760E 06 CONF LRC-SB



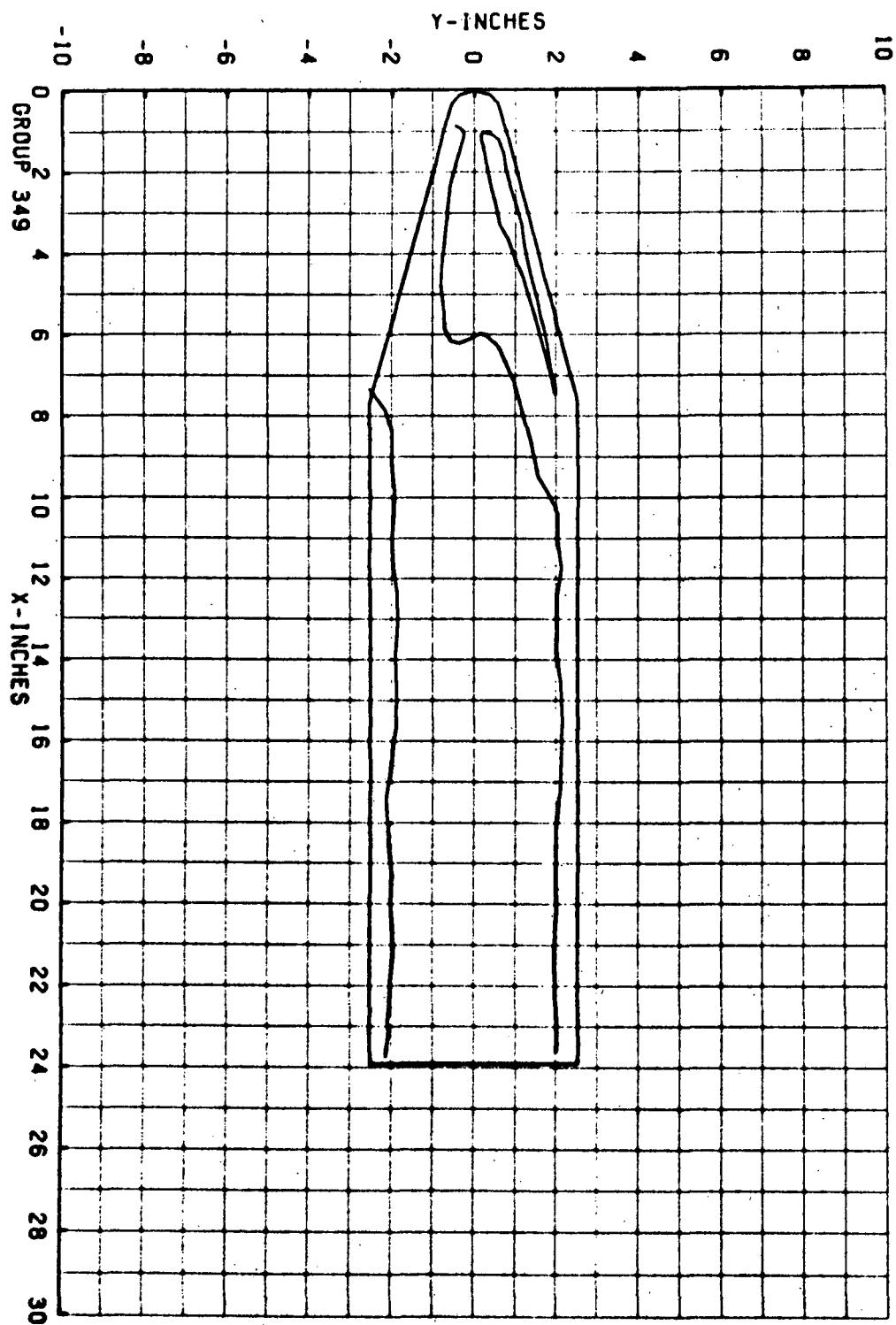
GROUP 349 PIC. NO. 871 H/HREF 6.120E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.768E-02 RE/FT 3.760E-06 CONF LRC-SB

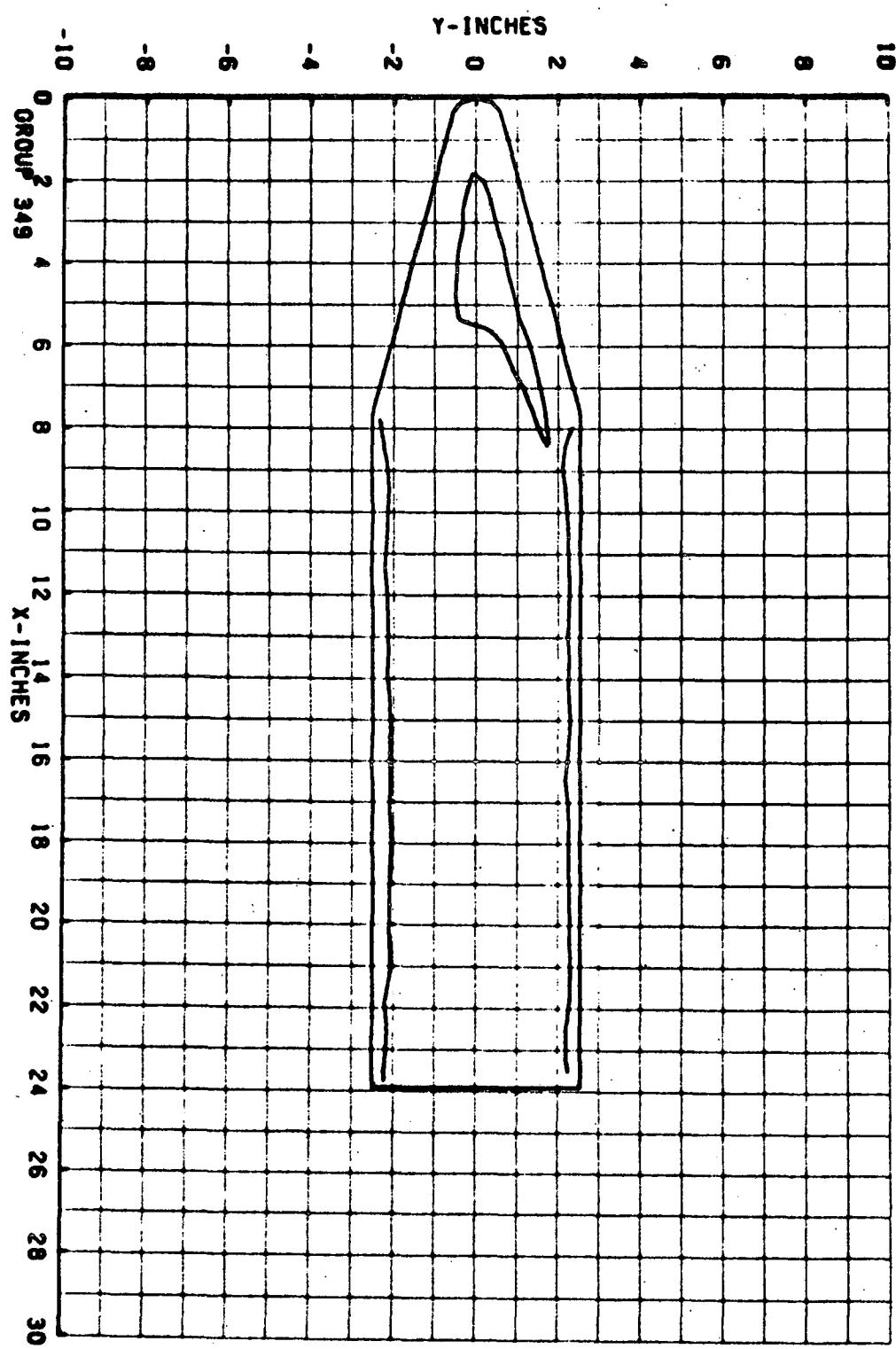


GROUP 349 PIC. NO. 872 H/HREF 5.566E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.768E-02 RE/FT 3.760E 06 CONF LRC-SB



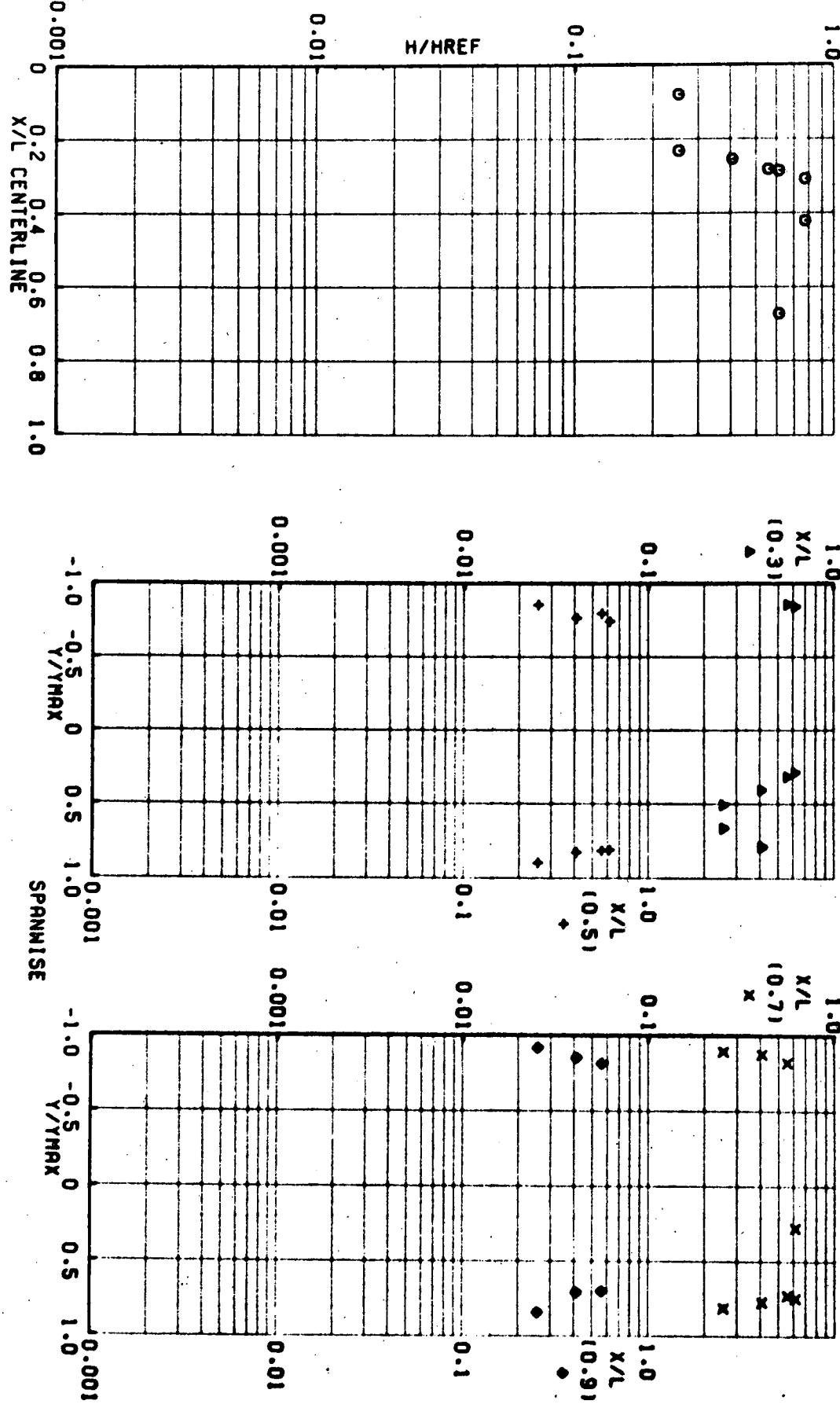
GROUP 349 PIC. NO. 877 H/HREF 4.036E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.768E-02 RE/FT 3.760E 06 CONF LRC-SB





GROUP 349 PIC. NO. 890 H/HREF 2.507E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.768E-02 RE/FT 3.760E 06 CONF LRC-SB

GROUP 349 ALPHA (DEG) 40.0 HREF 2.768E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FT 3.760E 06 CONF LRC-SB



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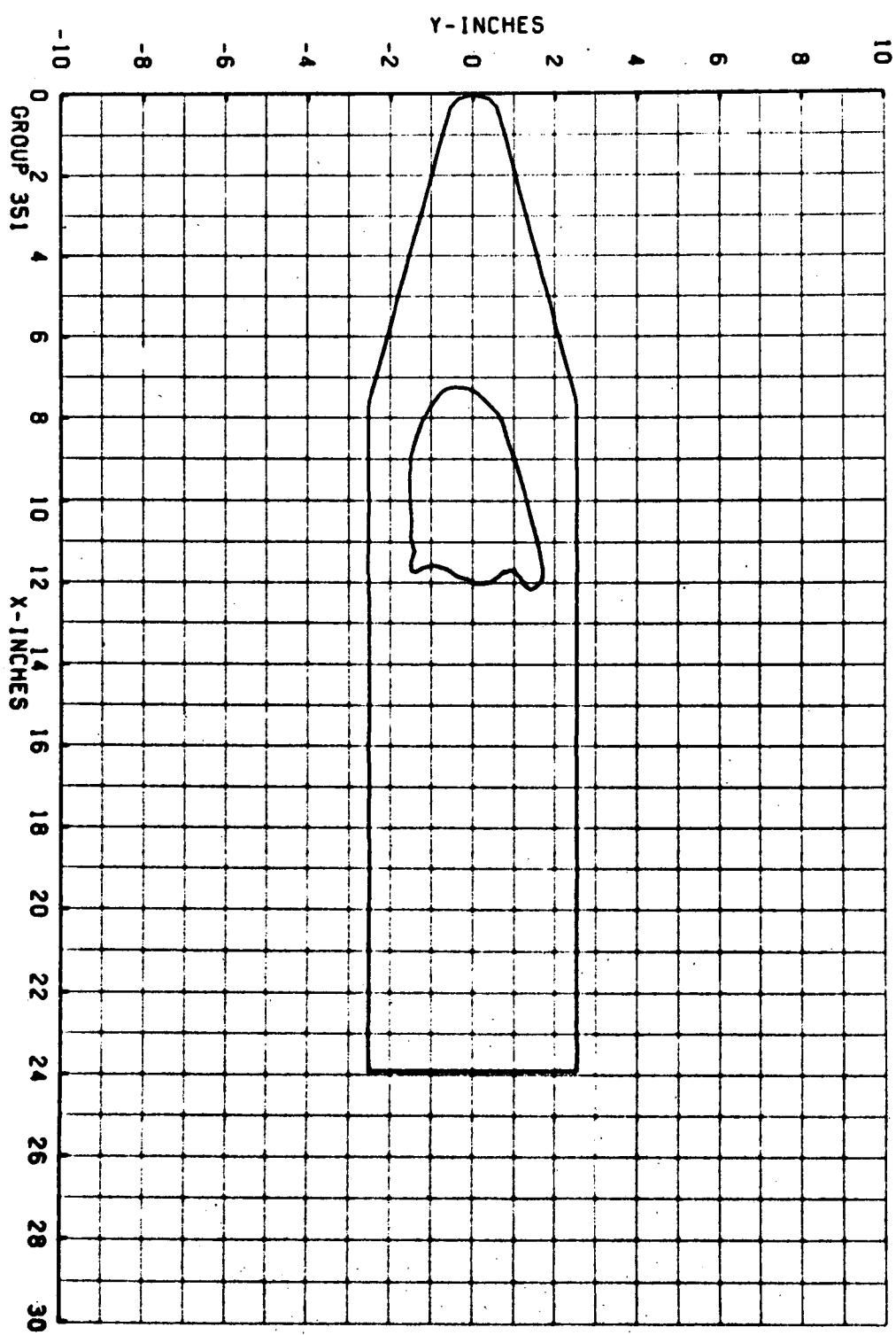
**A.F.D.C.(AMDO, INC.) ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL B**

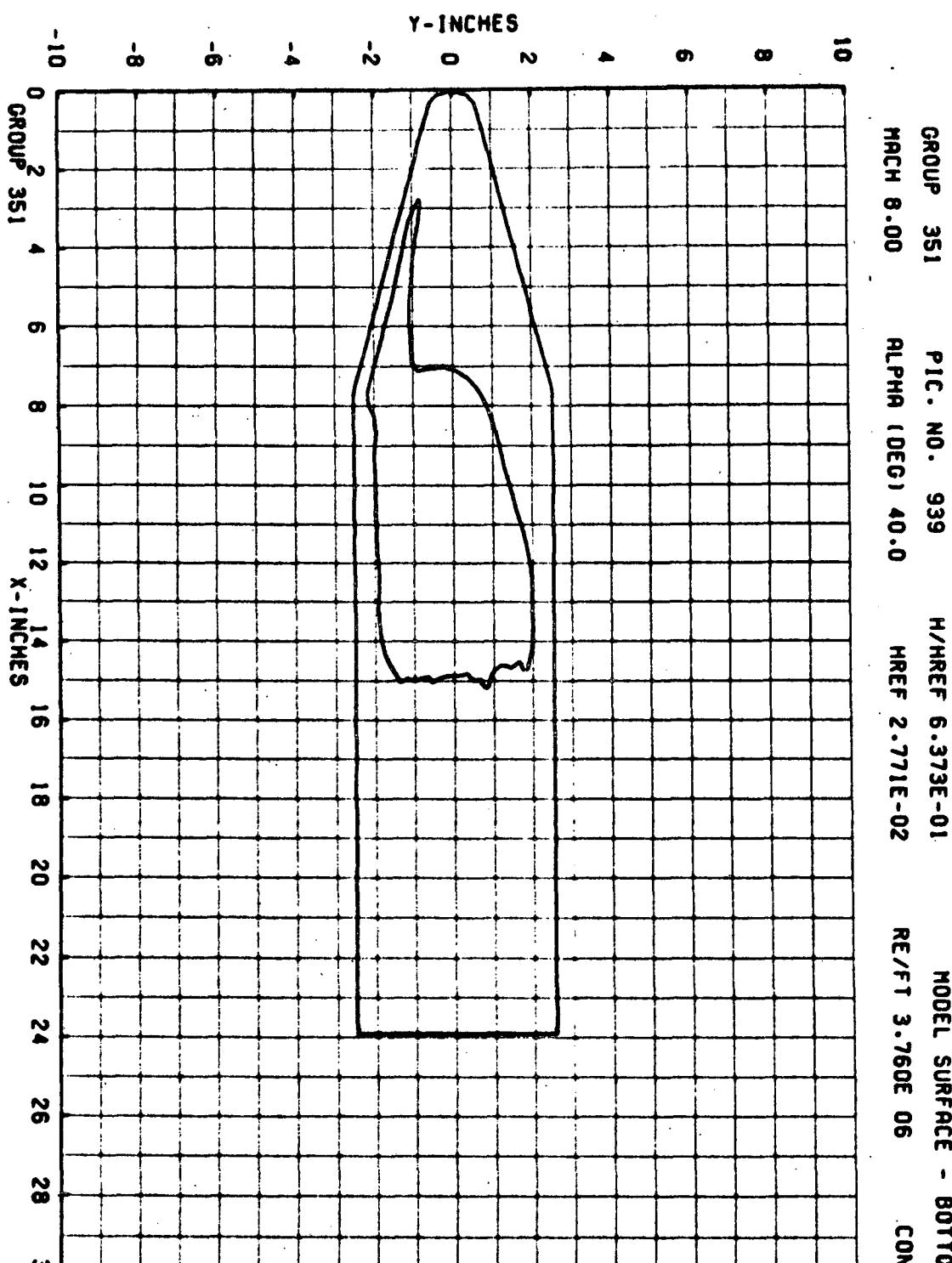
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GROUP-CONFIG-MODEL-MACH-N0-PROPSIA-TO-DEG-R-ALPHA-MODEL-ALPHA-SECTOR-ALPHA-PREHEND-ROLL-MODEL-VAN
351 12 LHC-SH A.00 863.0 1342 40.00 10.00 50.00 180.00 -0.0
1-INF 1-INF 1-INF V-INF RHO-INF MU-INF REFT HREF SIEF
(DEG R) (PSIA) (PSIA) (PSIA) (SLUGSF/SEC) (FT/SEC) (R=0.056FT) (W=0.056FT)
97.3 .000 3.960 .3866 7.624E-05 7.032E-08 3.76E-06 2.771E-02
CAMERA PAINT IEMP (DEG F) INITIAL TEMP (DEG F) SQUARE ROOT (RMUXCX)
TOP(1) 300
SIDE(S) 300 AVERAGE Tm = 96 = 0.008(SQUARE-ROOT DEL TIME) + 0.11
BOT(1)(B) 300

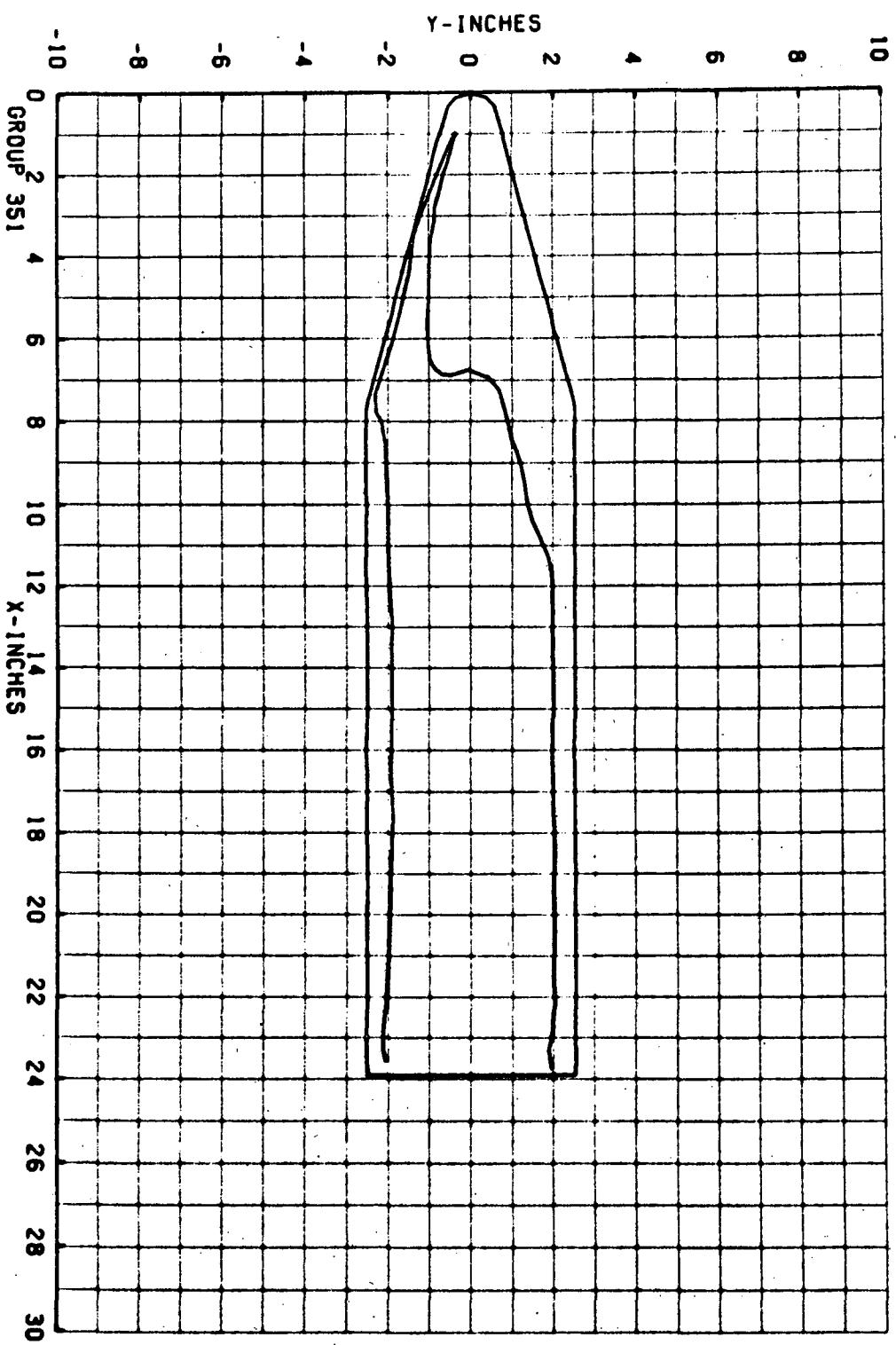
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GROUP 351 PIC. NO. 938 H/HREF 7.196E-01
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.771E-02
GROUP² 351 4 6 8 10 12 14 16 18 20 22 24 26 28 30 MODEL SURFACE - BOTTOM
CONF LRC-SB

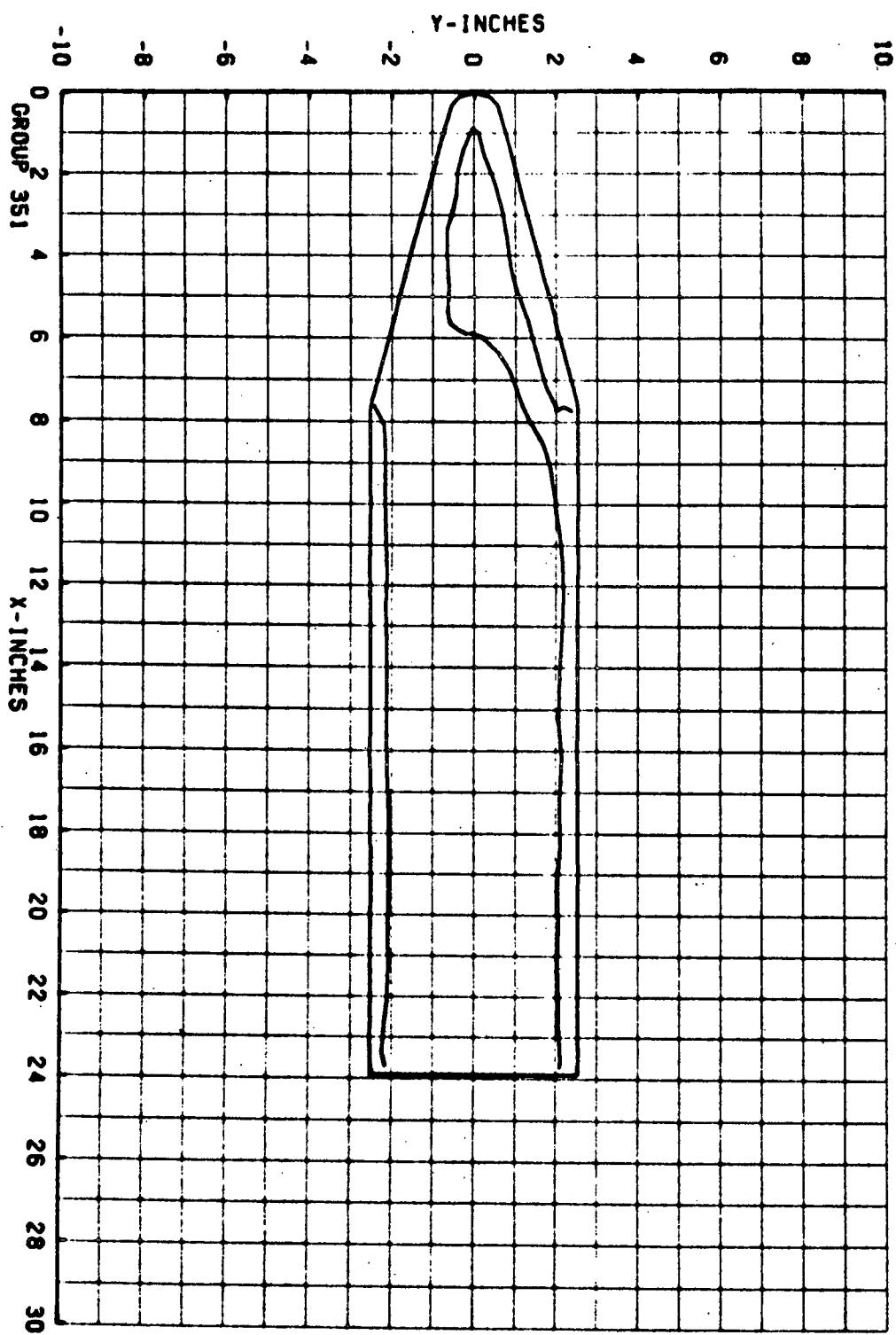




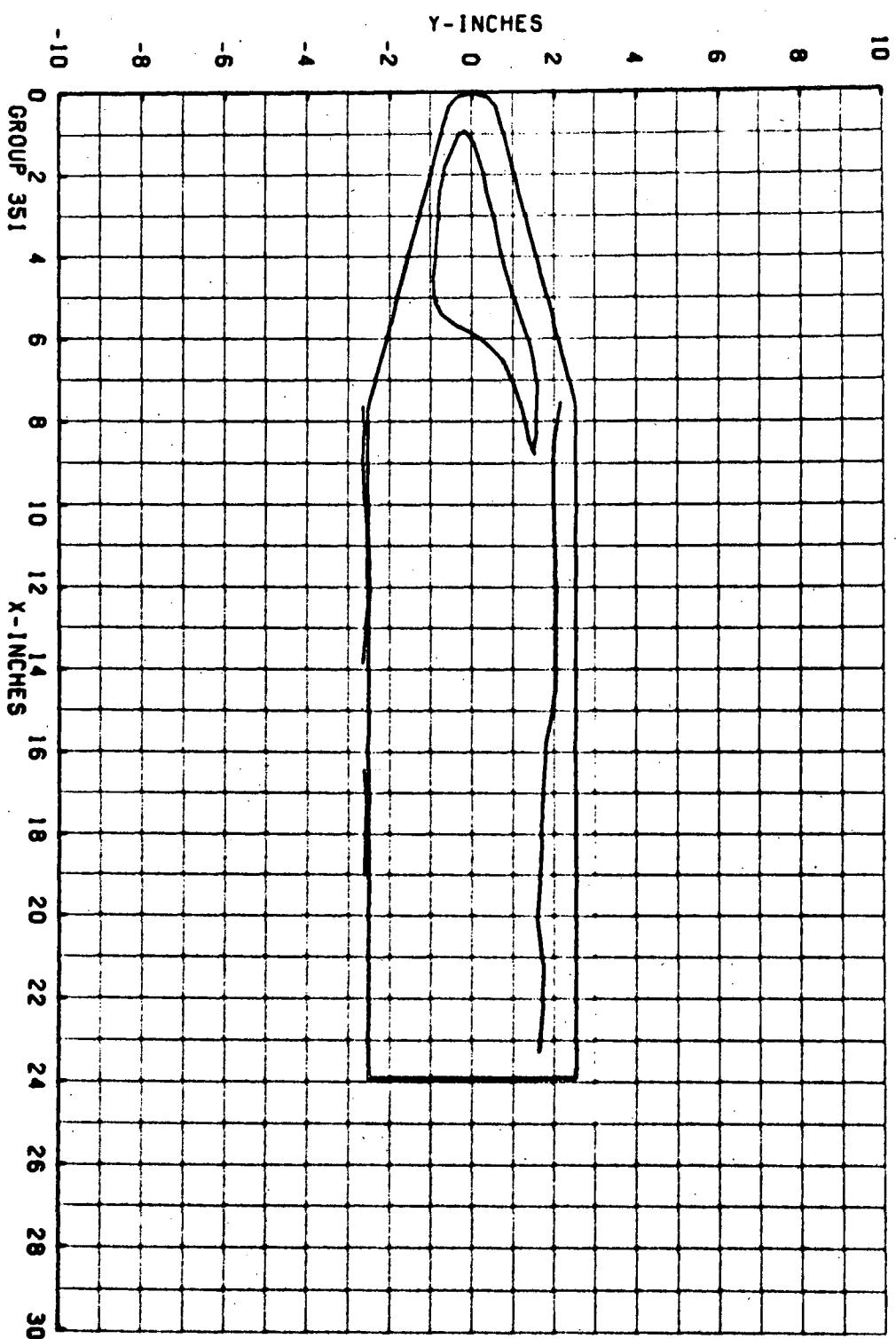
GROUP 351 PIC. NO. 940 M/HREF 5.709E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.771E-02 RE/FT 3.760E 06 CONF LRC-SB



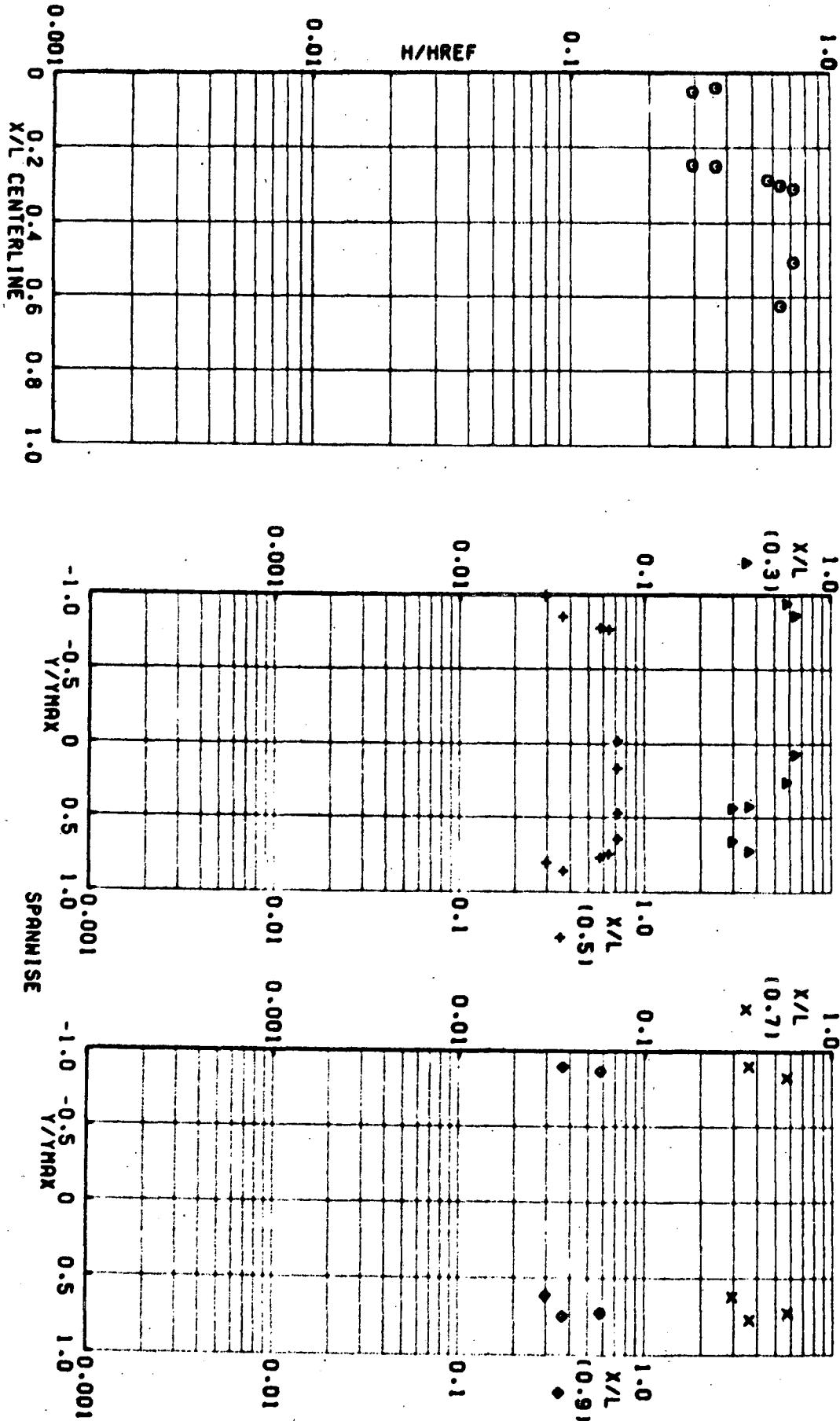
GROUP 351 PIC. NO. 947 H/HREF 3.591E-01 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 HREF 2.771E-02 RE/FT 3.760E 06 CONF LRC-SB



GROUP 351 PIC. NO. 952. MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 40.0 H/HREF 2.921E-01
HREF 2.771E-02 RE/FT 3.760E 06 CONF LRC-SB



GROUP 351 ALPHA (DEG) 40.0 MREF 2.771E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FT 3.760E 06 CONF LRC-S8



6/27/

AEDC (ANDO, INC.) ARNOLD AFS, TENNESSEE
VON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL A
VII162

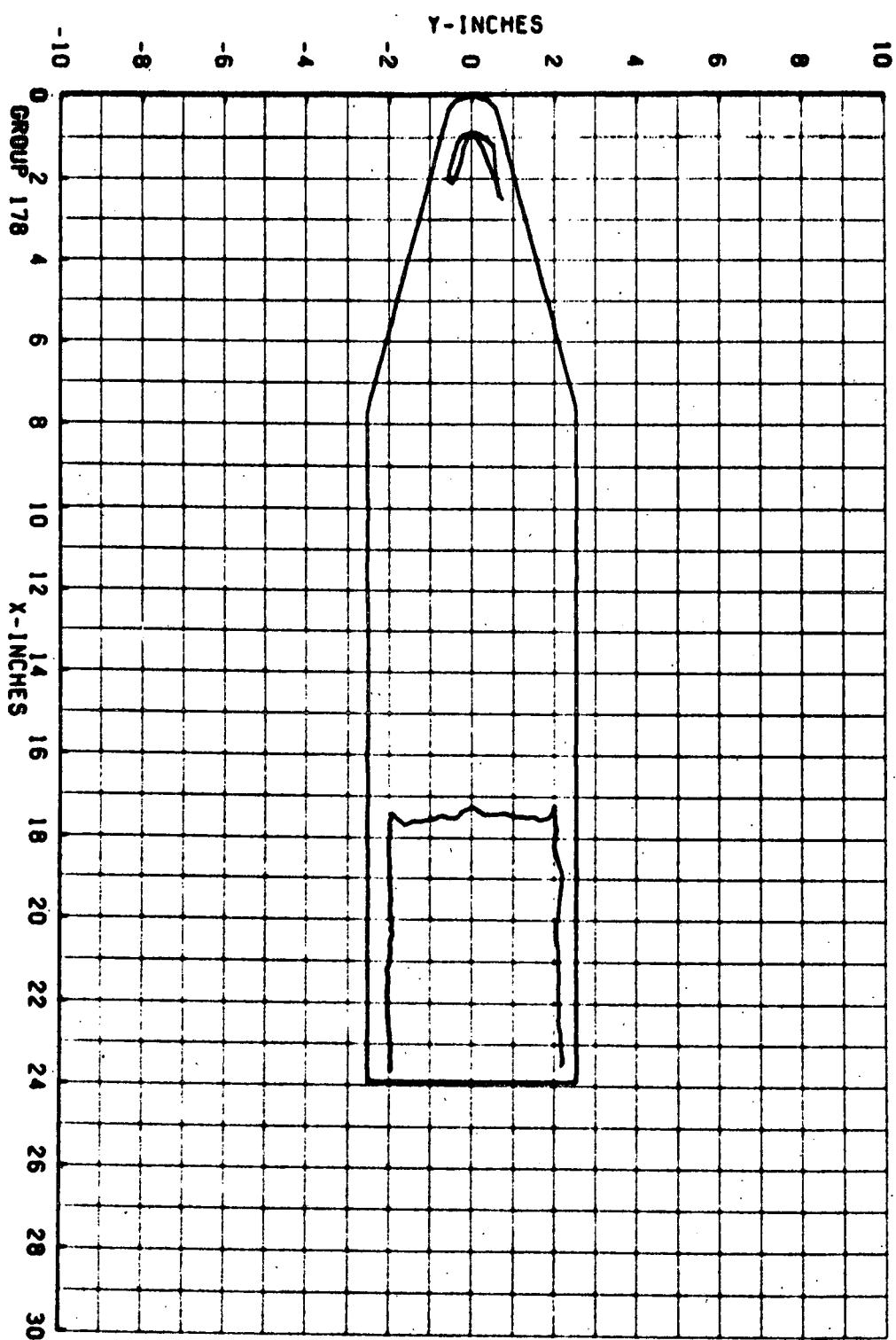
GROUP	CONFIG	MODEL	MACH NO	PROPSIA	T0 DEGR	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	VAV
178	12	LRC-SB	8.00	857.8	1349	59.98	-9.98	-50.00	180.00	0.0
	I-INF	P-INF	O-INF	V-INF	RHO-INF	MU-INF	REF	REF	STKEF	
	(DEG A)	(PSTA)	(VSTA)	(FT/SEC)	(SLURS/F3)	(BL-SG/F12)	(F1-1)	(Ra = 0.056FT)	(Ra = 0.056FT)	
	97.7	.058	3.936	3.875	7.533E-05	7.861E-08	3.72E 06	2.765E-02	1.175E-02	
	CAMERA	PAINT TEMP (DEG F)	INITIAL TEMP (DEG F)		SQUARE ROOT (RMOCRA)					
	TOP(1)	300								
	SIDE(S)	300								
	HOTWICH(1)									

AVERAGE TM = 91 -0.008(SQUARE.ROOT.DEL.TIME) + 0.11

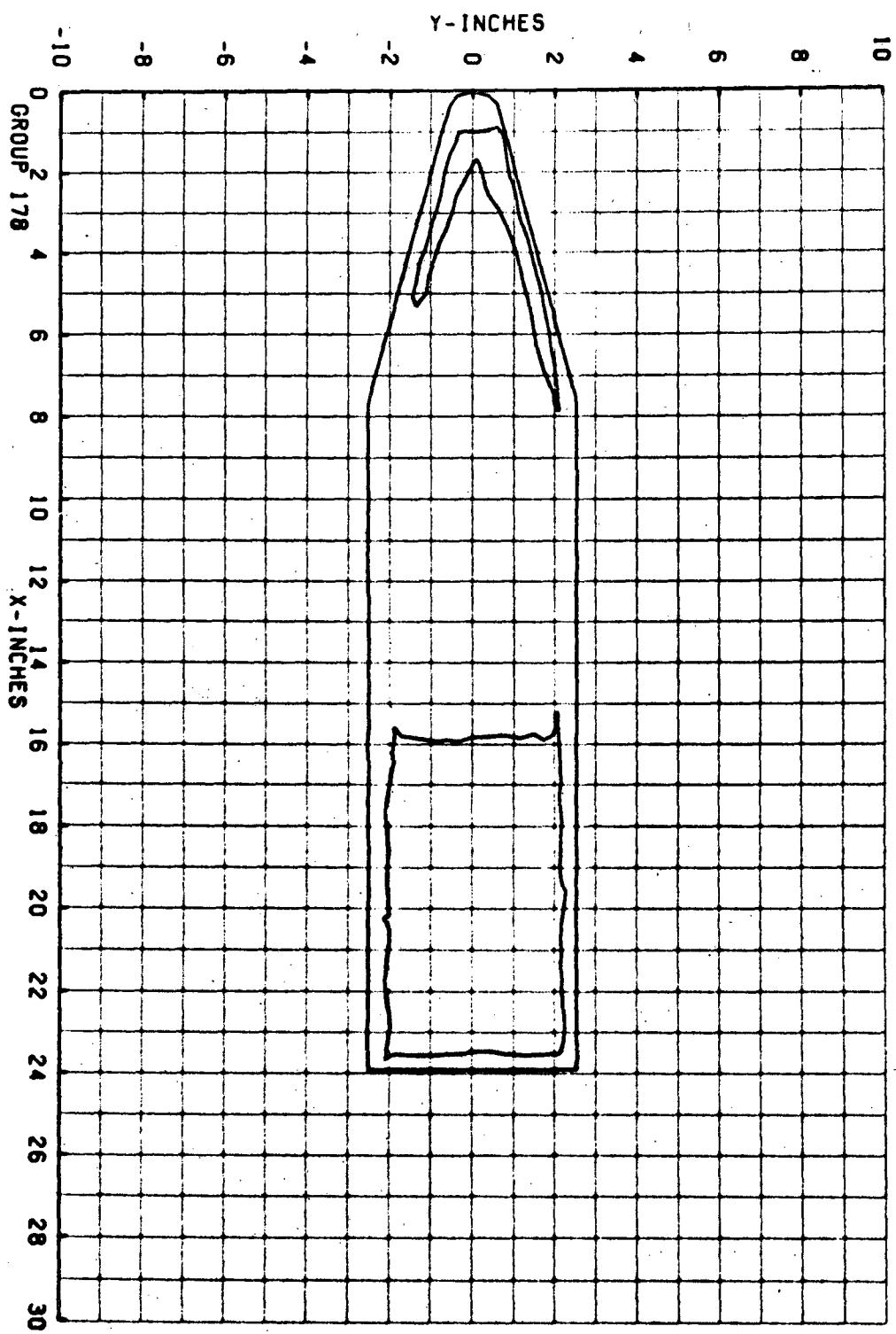
PTC NO	TYPE	REL TIME	HT01	HT02	HT03	HT04	HT05	HT06	HT07	HT08
1	3723	(300)	3.70	2.63	1.76E-02	.6371	2.250E-02	.8139	2.614E-02	.9456
1	3726	(300)	5.30	4.23	1.34E-02	.4844	1.711E-02	.6189	1.988E-02	.7190
1	3731	(300)	7.95	6.88	2.94E-03	.3615	1.277E-02	.6182	1.483E-02	.5365
1	3736	(300)	10.60	9.53	8.14E-03	.2945	1.040E-02	.3762	1.208E-02	.4371

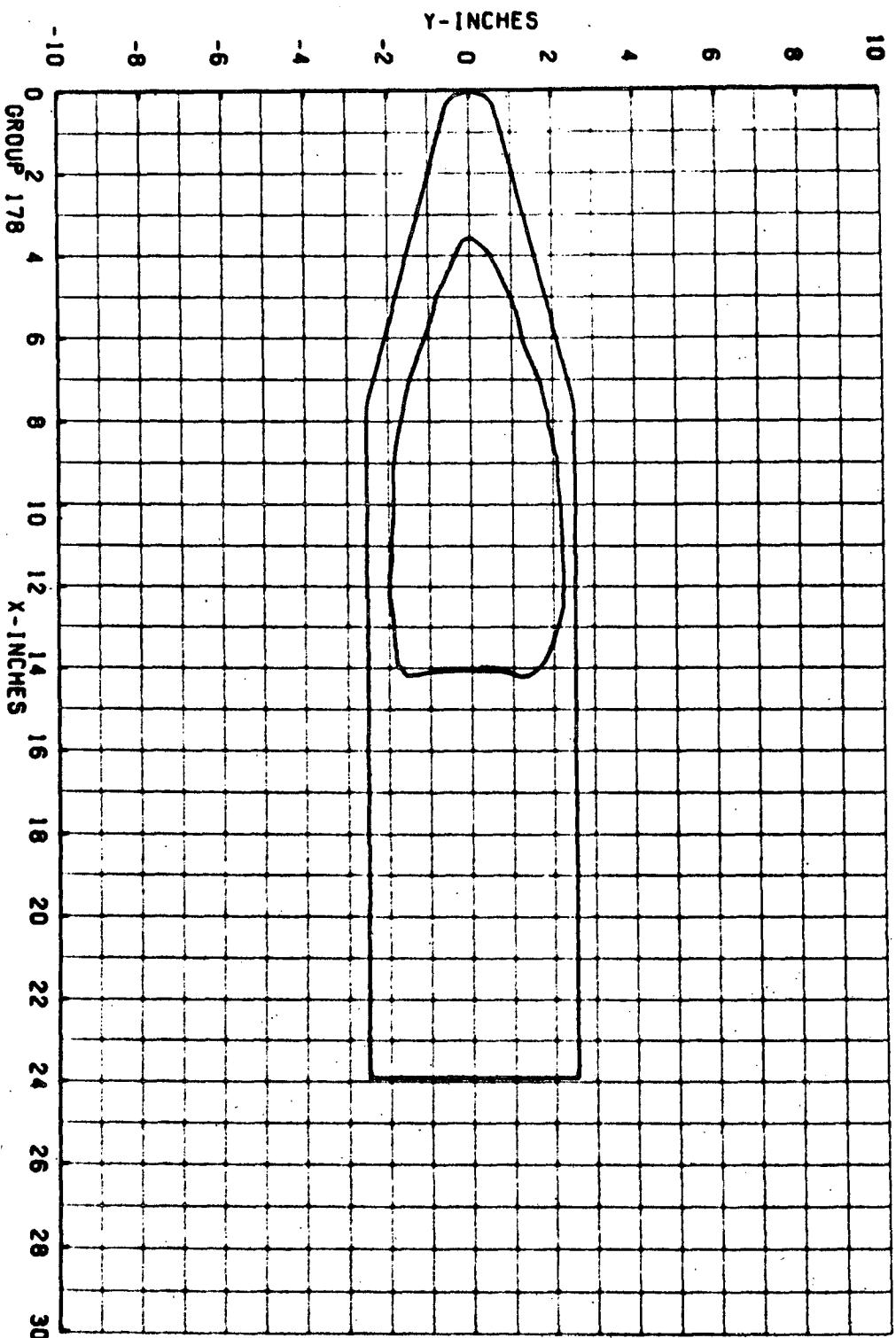
121

GROUP 178 PIC. NO. 3723 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 60.0 H/HREF 6.371E-01
ALPHA (DEG) 60.0 HREF 2.765E-02 RE/FI 3.720E 06
CONF LRC-SB

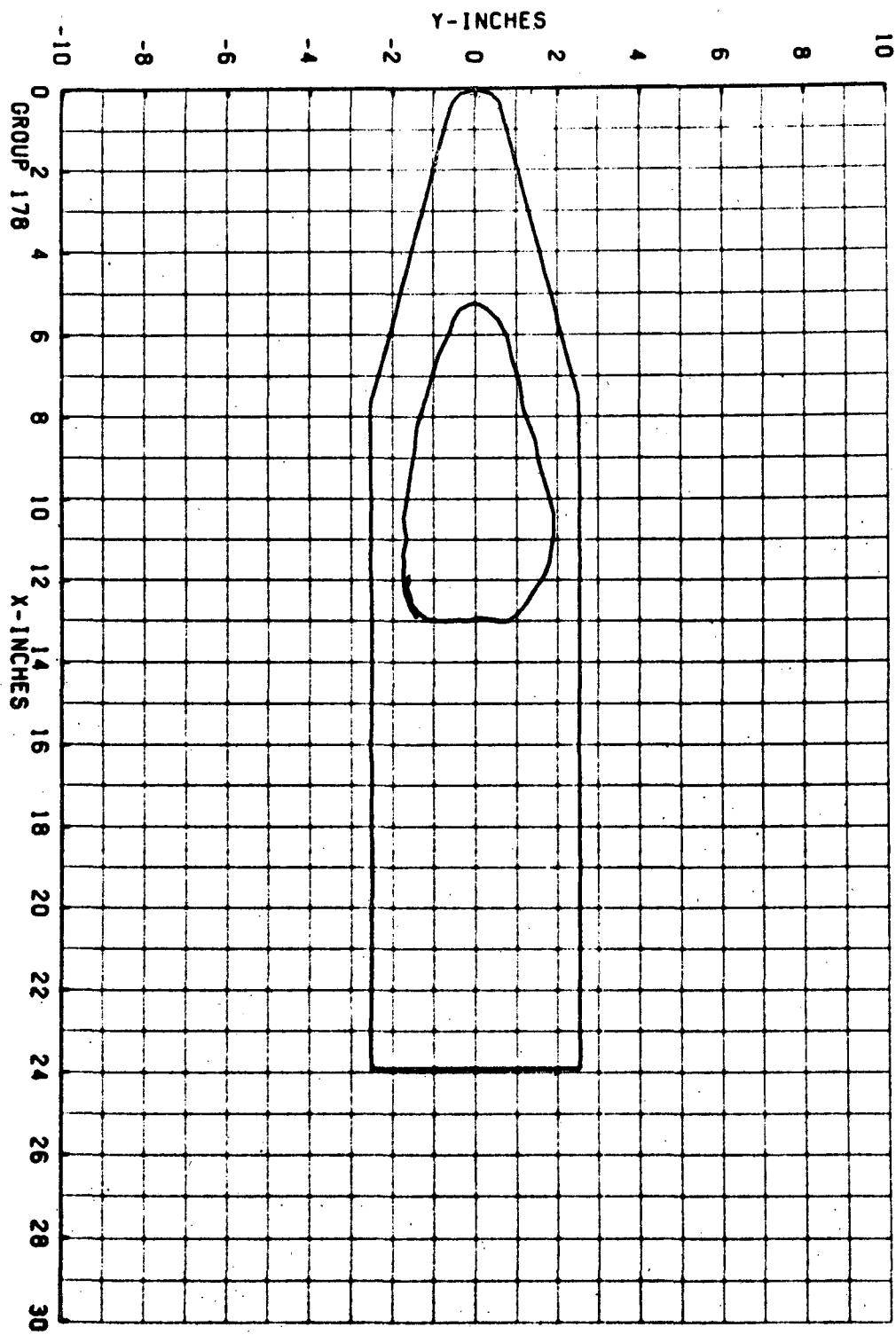


GROUP 178 PIC. NO. 3726 H/HREF 4.844E-01
MACH 8.00 ALPHA (DEG) 60.0 HREF 2.765E-02 RE/FT 3.720E 06 CONF LRC-S8
MODEL SURFACE - BOTTOM





GROUP 178 PIC. NO. 3736 MODEL SURFACE - BOTTOM
MACH 8.00 ALPHA (DEG) 60.0 H/HREF 2.945E-01
HREF 2.765E-02 RE/FT 3.720E 06 CONF LRC-SB



GROUP 178 ALPHA (DEG) 60.0 HREF 2.765E-02 MACH 8.00
 MODEL SURFACE - BOTTOM RE/FI 3.720E 06 CONF LRC-SB

