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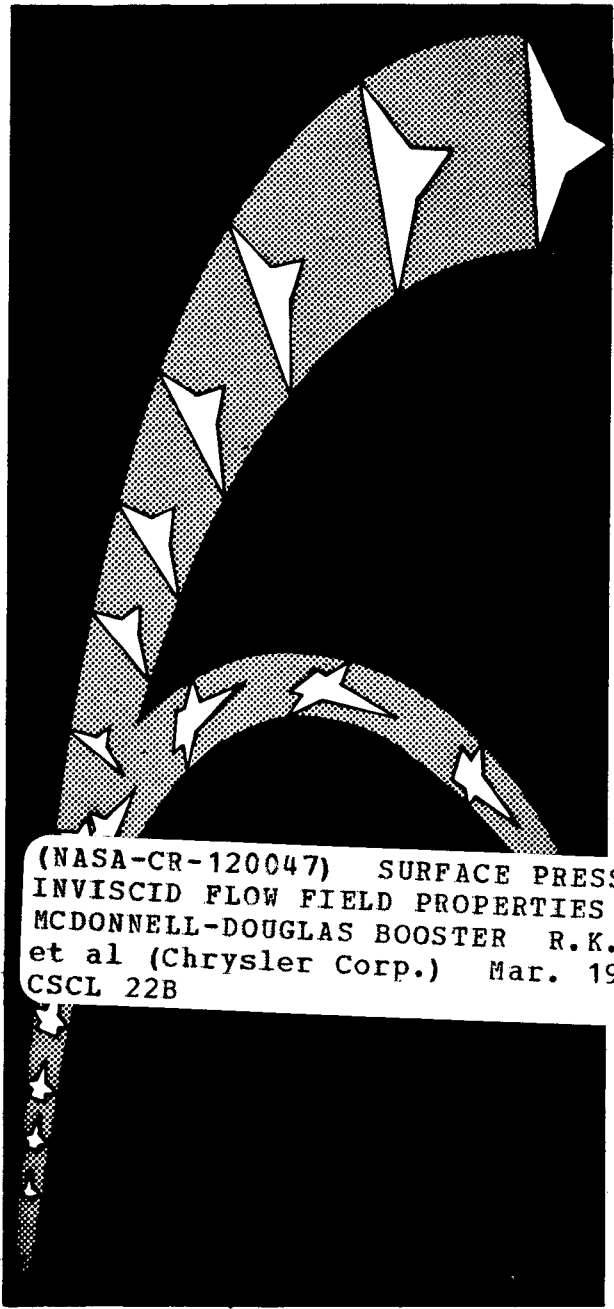
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CR-120,047  
VOLUME III  
MARCH 1972

—SPACE SHUTTLE—

**SURFACE PRESSURE AND INVISCID  
FLOW FIELD PROPERTIES  
MCDONNELL-DOUGLAS BOOSTER  
NOMINAL MACH NUMBER OF 8**

by

**R.K. Matthews, ARO, INC.  
W.R. Martindale, ARO, INC.  
J.D. Warmbrod, MSFC**



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INVISCID FLOW FIELD PROPERTIES  
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et al (Chrysler Corp.) Mar. 1972 52 p  
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VK F 50-INCH

**HYPERSONIC TUNNEL B**

**Arnold Engineering  
Development Center**

SADSAC SPACE SHUTTLE  
AEROTHERMODYNAMIC  
DATA MANAGEMENT SYSTEM

CONTRACT NAS8-4016  
MARSHALL SPACE FLIGHT CENTER



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VOLUME III  
March, 1972

SADSAC/SPACE SHUTTLE

WIND TUNNEL TEST DATA REPORT

CONFIGURATION: McDonnell-Douglas Booster

TEST PURPOSE: To Determine Surface Pressures and Inviscid Flow Field  
Properties at Mach Number 8

TEST FACILITY: AEDC VKF 50-Inch Hypersonic Tunnel B

TESTING AGENCY: AEDC-MSFC

TEST NO. & DATE: VT-1162-8; September, 1971

FACILITY COORDINATOR: Mr. L. L. Trimmer, ARO, INC.

PROJECT ENGINEER(S): Mr. R. K. Matthews, ARO, INC.  
Mr. W. R. Martindale, ARO, INC.  
Mr. J. D. Warmbrod, NASA-MSFC

DATA MANAGEMENT SERVICES

LIAISON: J. E. Vaughn DATA OPERATIONS: J. R. Ziler  
J. E. Vaughn J. R. Ziler

RELEASE APPROVAL: N. B. Kemp  
N. B. Kemp, Supervisor  
Aero Thermo Data Group

CONTRACT NAS 8-4016

AMENDMENT 153

DRL 184 - 58

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**FACILITY COORDINATOR:**

Mr. L. L. Trimmer, ARO, Inc.  
Arnold Engineering Development Center  
Arnold Air Force Station, Tennessee 37389

Phone: (615) 455-2611-X7377

**PROJECT ENGINEERS:**

Mr. R. K. Matthews ARO, Inc.  
Arnold Engineering Development Center  
Arnold Air Force Station, Tennessee 37389

Phone: (615) 455-2611-X594

Mr. W. R. Martindale, ARO, Inc.  
Arnold Engineering Development Center  
Arnold Air Force Station, Tennessee 37389

Phone: (615) 455-2611-X575

Mr. J. D. Warmbrod  
S&E - AERO - AF  
Building 4610  
NASA-MSFC  
Huntsville, Alabama 35812

Phone: (205) 453-0170

**SADSAC LIAISON:**

Mr. John E. Vaughn  
Chrysler Corp. - Huntsville Division  
102 Wynn Drive, Department 4820  
Huntsville, Alabama 35805

Phone: (205) 895-1560

**SADSAC OPERATIONS:**

Mr. J. R. Ziler  
Chrysler Corp. Space Division  
P. O. Box 29200, Department 2780  
New Orleans, Louisiana 70129

Phone: (504) 255-2304

## FOREWORD

The work reported herein was sponsored by the Marshall Space Flight Center (MSFC), NASA. The results of tests presented were obtained by ARO, Inc. (a subsidiary of Sverdrup & Parcel and Associates, Inc.), contract operator of the Arnold Engineering Development Center (AEDC), AFSC, Arnold Air Force Station, Tennessee. Ascent and reentry conditions were simulated on shuttle models designed by McDonnell Douglas (MDAC), North American Rockwell (NAR) and General Dynamics Convair (GDC). In addition a limited amount of data were obtained on two research models provided by the Langley Research Center (LRC). Because of the broad scope of these tests the data will be presented in a series of SADSAC reports. This report presents the results of the surface pressure and flow field tests conducted at Mach 8 in Tunnel B on the McDonnell Douglas Booster.

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## NOMENCLATURE

ALPHA-MODEL ( $\alpha$ )	Model angle of attack, deg
ALPHA-PREBEND	Sting prebend angle, deg
ALPHA-SECTOR	Tunnel sector angle, deg
CP	Pressure coefficient, $(P_M - (P - INF))/Q - INF$
CP-MAX	Pressure coefficient based on $P_{O1}$ , $(P_{O1} - (P - INF))/Q - INF$
L	Model length (23.78 in.)
MACH NO.	Free-stream Mach number
ML	Local Mach number
MU-INF	Free-stream viscosity, lb-sec/ft <sup>2</sup>
MUL	Local viscosity, lb-sec/ft <sup>2</sup>
P-INF	Free-stream pressure, psia
PM	Model surface pressure, psia
PML	Local model surface pressure, psia
P0	Tunnel stilling chamber pressure, psia
P01	Stagnation pressure downstream of a normal shock, psia
PR	Rake probe stagnation pressure, psia
Q-INF	Free-stream dynamic pressure, psia
RE/FT	Free-stream unit Reynolds number, ft <sup>-1</sup>
REL	Local unit Reynolds number, ft <sup>-1</sup>
RHO-INF	Free-stream density, LBM/ft <sup>3</sup>
RHOL	Local density, LBM/ft <sup>3</sup>
RHOUL	Local density-velocity product, LBM/ft <sup>2</sup> -sec
ROLL-MODEL ( $\phi$ )	Model roll angle, deg
T-INF	Free-stream temperature, °R
TL	Local temperature, °R
TO	Tunnel stilling chamber temperature, °R

TTR	Total temperature measured by rake probes, °R
U-INF	Free-stream velocity, ft/sec
UL	Local velocity, ft/sec
X	Axial coordinate (see Fig. 1), in.
Y	Distance from model surface or probe height (see Figs. 1 and 3), in.
YAW	Model yaw angle, deg.



## SECTION 1

### INTRODUCTION

This report presents the results of a wind tunnel test program to determine surface pressures and flow field properties on the McDonnell Douglas Booster configuration. The tests were conducted at the Arnold Engineering Development Center (AEDC) in Tunnel B of the von Karman Gas Dynamics Facility (VKF). The tests were conducted in September 1971.

Data were obtained at a nominal Mach number of 8 at angles of attack of 40- and 50-deg and at a free-stream unit Reynolds number of  $3.7 \times 10^6$  per foot.

## SECTION 2

### MODELS AND APPARATUS

#### 2.1 MODEL DESCRIPTION

Model drawings were provided ARO, Inc. by the McDonnell Douglas Corporation and fabrication of the Stycast model was subcontracted to the Grumman Aircraft Corporation. The model had 10 windward centerline orifices. A sketch showing the overall model dimensions is presented in Fig. 1 and a photograph of the configuration is shown in Fig. 2. Table 1 provides additional configuration description details but it should be pointed out that the model was cast as one smooth surface without moveable control surfaces.

#### 2.2 FACILITY DESCRIPTION

Tunnel B is a continuous, closed-circuit, variable density wind tunnel with an axisymmetric contoured nozzle and a 50-in.-diam test section.

The tunnel can be operated at a nominal Mach number of 6 or 8 at stagnation pressures from 20 to 300 and 50 to 900 psia, respectively, and at stagnation temperatures up to 1350°R. The model can be injected into the tunnel for a test run and then retracted for model cooling or model changes without interrupting the tunnel flow.

### 2.3 INSTRUMENTATION

The model flow field was surveyed with pitot-pressure and single shield total temperature probe rakes. The rakes were mounted so that pressure and temperature measurements could be made simultaneously. The rakes, support mechanism, and spacing of the probes are shown in Fig. 3.

Static and pitot-probe pressures were measured with 15 psid transducers referenced to a near vacuum for pressures less than 15 psia and to atmospheric pressure for pressures greater than 15 psia. The atmospheric reference pressure was also measured with a 15 psid transducer.

## SECTION 3

### PROCEDURE

#### 3.1 TEST CONDITIONS

Nominal test conditions are presented in the data summary sheets (Table 2) and the specific test conditions for each run (or group) are provided at the top of the data tabulation sheet for that run.

### 3.2 DATA REDUCTION

By assuming the flow-field static pressure equal to the wall static pressure, the local Mach number (ML) was calculated from the Rayleigh pitot formula,

$$\frac{P_R}{P_{ML}} = \left( \frac{6ML}{5} \right)^{7/2} \left( \frac{6}{7ML^2 - 1} \right)^{5/2}, \text{ for } ML \geq 1$$

or from the compressible Bernoulli equation,

$$\frac{P_R}{P_{ML}} = (1 + 0.2 ML^2)^{7/2}, \text{ for } ML < 1.$$

The assumption of constant static pressure becomes less valid as the distance from the model surface increases.

The equations for the other flow field parameters are:

<u>Parameter</u>	<u>Equation</u>	<u>Units</u>
TL	$TL = \frac{T_0}{(1 + 0.2 ML^2)}$	°R
UL	$UL = (49.02)(ML) \sqrt{TL}$	ft/sec
RHOL	$RHOL = \frac{(2.70)(P_{ML})}{TL}$	LBM/ft <sup>3</sup>
MUL	$MUL = \frac{2.27 (TL)^{3/2}}{TL + 198.6} \times 10^{-8}$	lb-sec/ft <sup>2</sup>
REL	$REL = \frac{(RHOL)(UL)}{(32.17)(MUL)}$	ft <sup>-1</sup>

The quantities calculated using TL are not valid in the model boundary layer since TTR is less than T0 and, of course, none of the calculated parameters are meaningful outside the model shock layer.

### 3.3 DATA PRECISION

Estimated uncertainties of the primary measurements are given below:

<u>Parameter</u>	<u>Uncertainty</u>
PML	$\pm 0.015$ psia
P0	$\pm 1.8$ psia
P01	$\pm 0.021$ psia
PR	$\pm 0.015$ psia (for PR $\leq$ 15 psia) $\pm 0.021$ psia (for PR $>$ 15 psia)
T0	$\pm 10^\circ\text{R}$
TTR	$\pm 25^\circ\text{R}$

## SECTION 4

### DATA PRESENTATION

The test data are presented in tabulated and plotted form in Appendix A with the corresponding shadowgraph picture following the plotted data. The data are presented in the following order.

<u>Appendix</u>	<u>Type Data</u>	<u><math>\alpha</math></u>	<u>X/L</u>
A ↓	Surface	40	0.1 → 0.97
	Pressure	50	↓
	Flow	40	0.3, 0.5, 0.7, 0.9
	Field	50	↓

Table 3, Page 15, presents a summary of these data.

Pitot pressure and total temperature measurements were attempted at 60 degrees angle of attack; however, the rakes and support distorted the flow field as observed in shadowgraph photographs and therefore these measurements are not presented.

The total temperature probes were quite delicate and subject to failure. In cases where the probes failed and could not be replaced immediately the measurement does not appear in the data tabulation or on the plot.

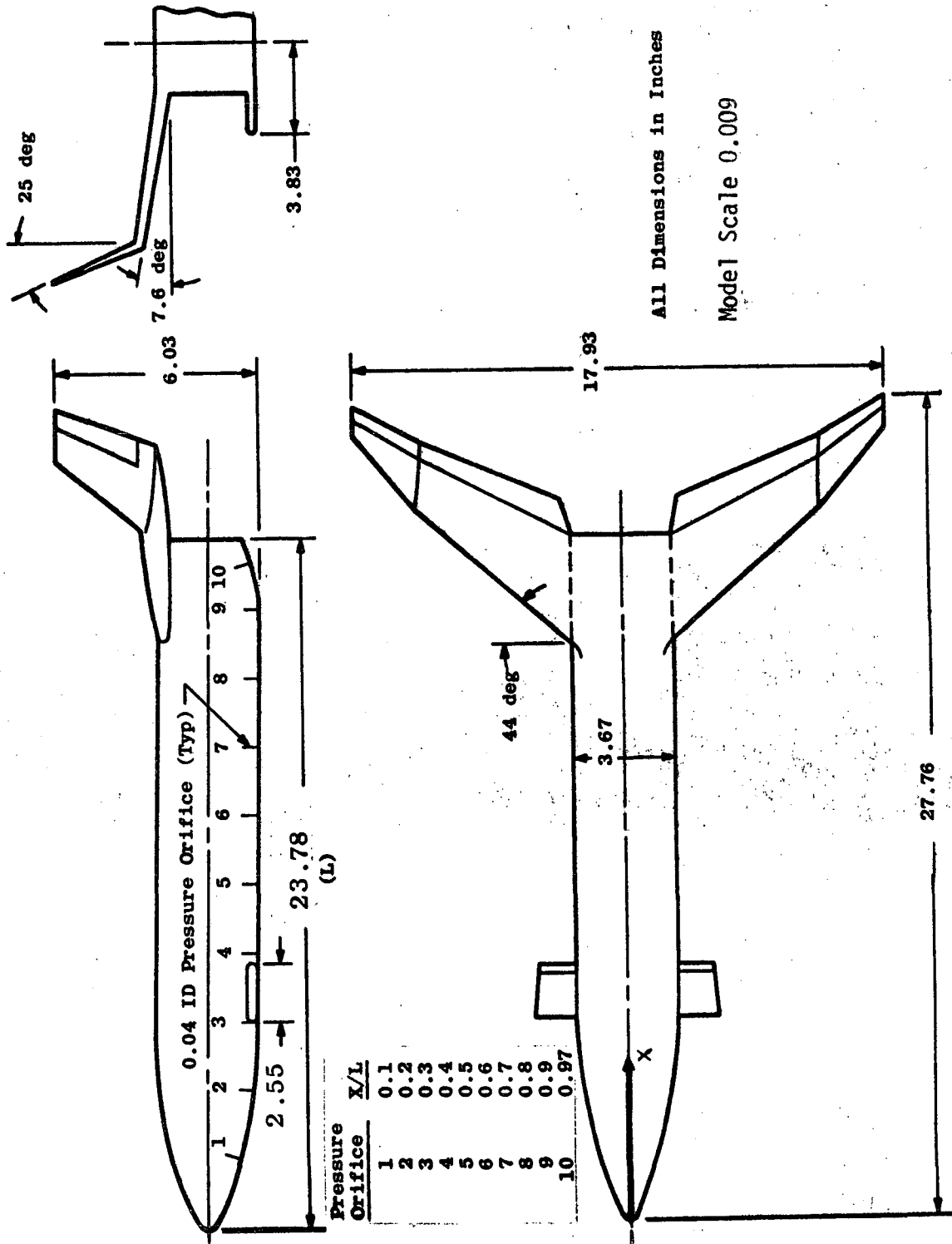


Figure 1. McDonnell-Douglas Booster Model Sketch (0.009 Scale)

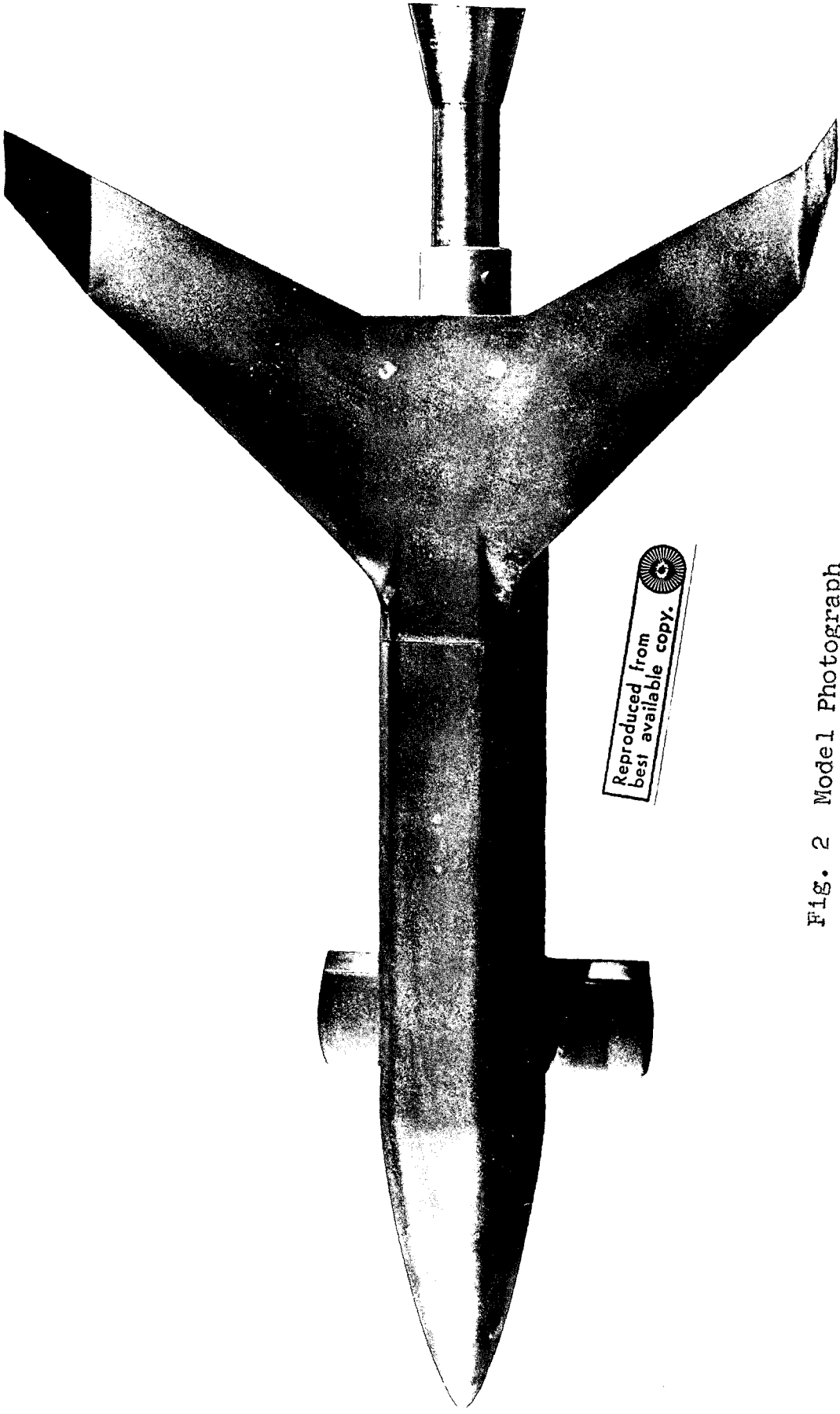


Fig. 2 Model Photograph

Probe Height, Y, in.		
No.	Pressure Probes	Temperature Probes
1	0.014	0.051
2	0.066	0.131
3	0.112	0.202
4	0.163	0.303
5	0.216	0.402
6	0.258	0.599
7	0.313	
8	0.365	
9	0.415	
10	0.499	
11	0.606	
12	0.702	
13	0.802	
14	0.892	
15	0.981	

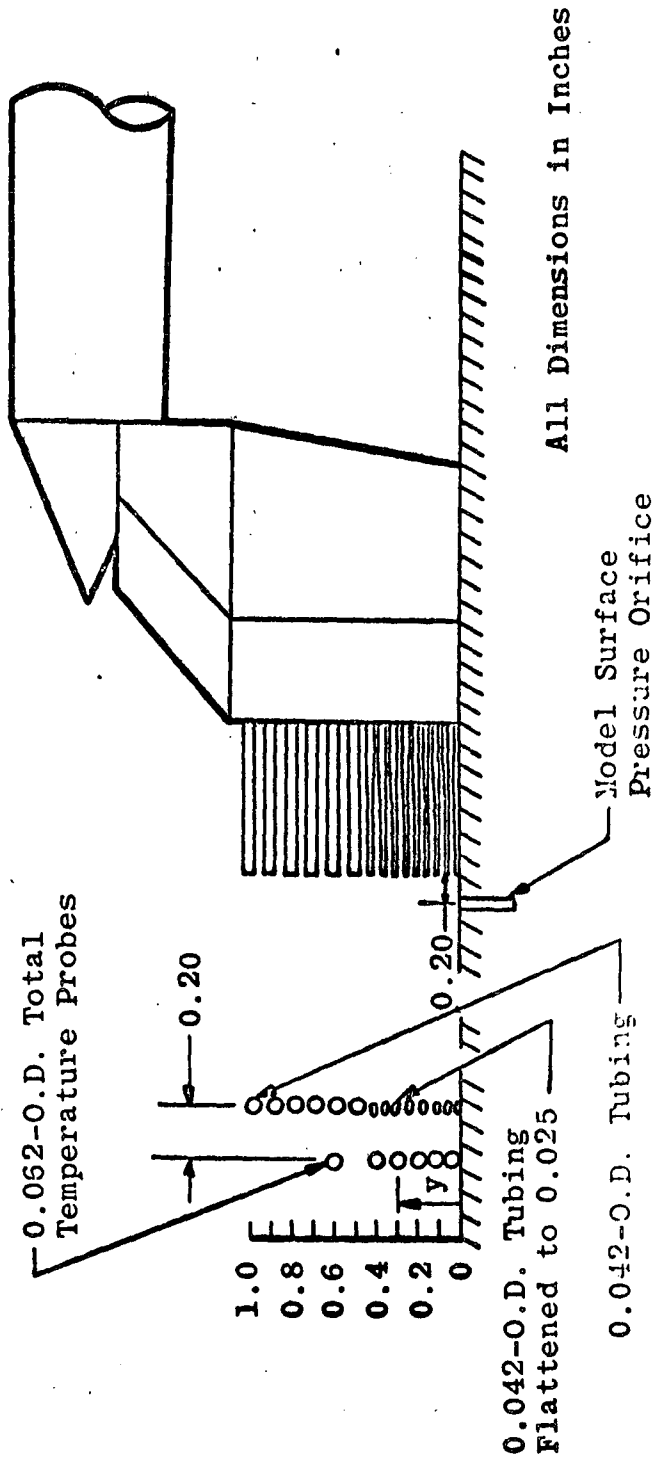
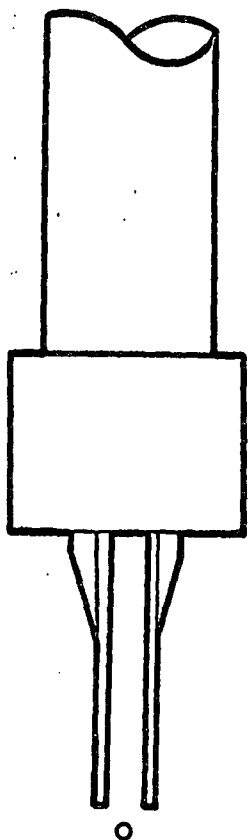


Fig. 3 Probe-Rakes and Support



Table 1  
Configuration Description Details

**MODEL COMPONENT:** BODY - MDAC Booster

**GENERAL DESCRIPTION:** Configuration 256-17A booster; model scale 0.009

**DRAWING NUMBER:** 256-17-0001, Rev. A

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length (ft)	<u>220.17</u>	<u>1.98</u>
Max. Width (ft)	<u>34.0</u>	<u>0.306</u>
Max. Depth (ft)	<u>34.0</u>	<u>0.306</u>
Fineness Ratio	<u>          </u>	<u>          </u>
Area		
Max. Cross-Sectional	<u>          </u>	<u>          </u>
Planform	<u>          </u>	<u>          </u>
Wetted	<u>          </u>	<u>          </u>
Base	<u>          </u>	<u>          </u>

TABLE 1 - CONTINUED

MODEL COMPONENT: Wing - MDAC Booster

GENERAL DESCRIPTION: Configuration 17A Wing

Model Scale 0.009

DRAWING NUMBER: \_\_\_\_\_

DIMENSIONS: FULL-SCALE MODEL SCALE

TOTAL DATA

Area, ft <sup>2</sup>		
Planform	6020.0	.488
Wetted		
Span (equivalent), ft.	146.0	1.314
Aspect Ratio	3.54	3.54
Rate of Taper		
Taper Ratio	.435	.435
Diehedral Angle, degrees	7.67	7.67
Incidence Angle, degrees	3.0	3.0
Aerodynamic Twist, degrees	0	0
Toe-In Angle		
Cant Angle		
Sweep Back Angles, degrees		
Leading Edge	44.0	44.0
Trailing Edge		
0.25 Element Line		
Chords:		
Root (Wing Sta. 0.0), inches	690.0	6.21
Tip, (equivalent)	300.0	2.70
MAC, inches	520.0	4.68
Fus. Sta. of .25 MAC, inches		
W.P. of .25 MAC, inches		
Airfoil Section		
Root	0010-64	0010-64
Tip	0010-64	0010-64

EXPOSED DATA

Area, ft <sup>2</sup>	4190.0	.339
Span, (equivalent), ft.		
Aspect Ratio		
Taper Ratio		
Chords		
Root, inches	594.0	5.35
Tip, inches	300.0	2.70
MAC		
Fus. Sta. of .25 MAC		
W.P. of .25 MAC		

TABLE 1 - CONTINUED

MODEL COMPONENT: Vertical Tails - MDAC Booster

GENERAL DESCRIPTION: Configuration 17 Vertical Tails

Model Scale 0.009

DRAWING NUMBER: 256-17-0001, Rev. A

DIMENSIONS: FULL-SCALE MODEL SCALE

TOTAL DATA (Values for one)

Area		
Planform (True)	438	.035
(Side Projection)	397	.032
Span (equivalent), inches	276	2.48
Aspect Ratio	1.21	1.21
Rate of Taper		
Taper Ratio	.520	.520
Dihedral Angle, degrees		
Incidence Angle, degrees		
Aerodynamic Twist, degrees		
Toe-In Angle	0	0
Cant Angle	25	25
Sweep Back Angles, degrees		
Leading Edge	40	40
Trailing Edge		
0.25 Element Line		
Chords:		
Root (Wing Sta. 0.0)	300	2.70
Tip, (equivalent), inches	156	1.40
MAC, inches	236	2.12
Fus. Sta. of .25 MAC		
W.P. of .25 MAC		
Airfoil Section		
Root	NACA 64A-009	NACA 64A-009
Tip	NACA 64A-009	NACA 64A-009

EXPOSED DATA

Area		
Span, (equivalent)		
Aspect Ratio		
Taper Ratio		
Chords		
Root		
Tip		
MAC		
Fus. Sta. of .25 MAC		
W.P. of .25 MAC		

TABLE 1 - CONTINUED

MODEL COMPONENT: Elevon - MDAC Booster

GENERAL DESCRIPTION: Configuration 17A Elevons

Model Scale 0.009

DRAWING NUMBER: 256-17-0001, Rev. A

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area	<u>617 ft<sup>2</sup></u>	<u>.0499 ft<sup>2</sup></u>
Span (equivalent)	<u>650 in.</u>	<u>5.85 in.</u>
Inb'd equivalent chord	<u>180 in.</u>	<u>1.62 in.</u>
Outb'd equivalent chord	<u>93 in.</u>	<u>.837 in.</u>
Ratio Elevator chord/horizontal tail chord		
At Inb'd equiv. chord	<u>.3</u>	<u>.3</u>
At Outb'd equiv. chord	<u>.3</u>	<u>.3</u>
Sweep Back Angles, degrees		
Leading Edge	<u>33</u>	<u>33</u>
Tailing Edge	<u>27</u>	<u>27</u>
Hingeline	<u>33</u>	<u>33</u>
Area Moment (Normal to hinge line), ft <sup>3</sup>	<u>2998</u>	<u></u>

TABLE 1 - CONCLUDED

MODEL COMPONENT: Canard - MDAC Booster

GENERAL DESCRIPTION: Configuration 17A Canard

Model Scale 0.009

DRAWING NUMBER: 256-17-001, Rev. A

DIMENSIONS:

	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Theo. Area, ft <sup>2</sup>	1660	.134
Exp Area, ft <sup>2</sup>	1215	.098
Aspect Ratio	3.0	3.0
Chord (Incl. Flap), ft	23.625	.213
Airfoil (360 In. Theo Chord)	NACA 63-018	NACA 63-018

Table 2

Test Data Summary Sheet

TEST TITLE: MDAC-Booster Flow Field Tests

TEST NUMBER: VT1162

TEST FACILITY: AEDC Tunnel B

TEST DATE: September 1971

TEST ENGINEER: R. K. Matthews & W. R. Martindale

Run No.	Model Configuration Identification	Model Scale	Free Stream Mach Number	Total Pressure (psia)	Total Temp. (°R) x 10 <sup>-6</sup>	Re/ft	Flow Field Survey Station X/I.	Type: Data#	Model Position (degrees)		Remarks	
									$\alpha$	$\phi$		
335	MDAC-B	0.009	8	850	1345	3.7	N/A	SP	40	0	180	
336							N/A	"	50			
339							.3	FF	40			
337							.5					
333							.7					
331							.9					
340							.3		50			
338							.5					
334							.7					
332							.9					

\*SP - Surface Pressure  
FF - Flow Field

TABLE 3

SUMMARY DATA PLOT INDEX

TYPE OF DATA	PAGES	ANGLE OF ATTACK - DEGREES		FLOW FIELD SURVEY STATION (X/L)				
		40	50	0.3	0.5	0.7	0.9	
<u>SURFACE PRESSURE</u> ↓	18 21	X	X	N/A				
<u>FLOW FIELD</u> ↓	24 27 30 33 36 39 42 45	X X X X	X  X X X X	X  X	X  X	X  X	X  X	

SURFACE PRESSURE

PM/POI vs. X/L

PR/POI vs. Y

TTR/TO vs. Y

ML vs. Y

UL/U-INF vs. Y

RHOUL/RHOI-INF vs. Y

FLOW FIELD

APPENDIX A

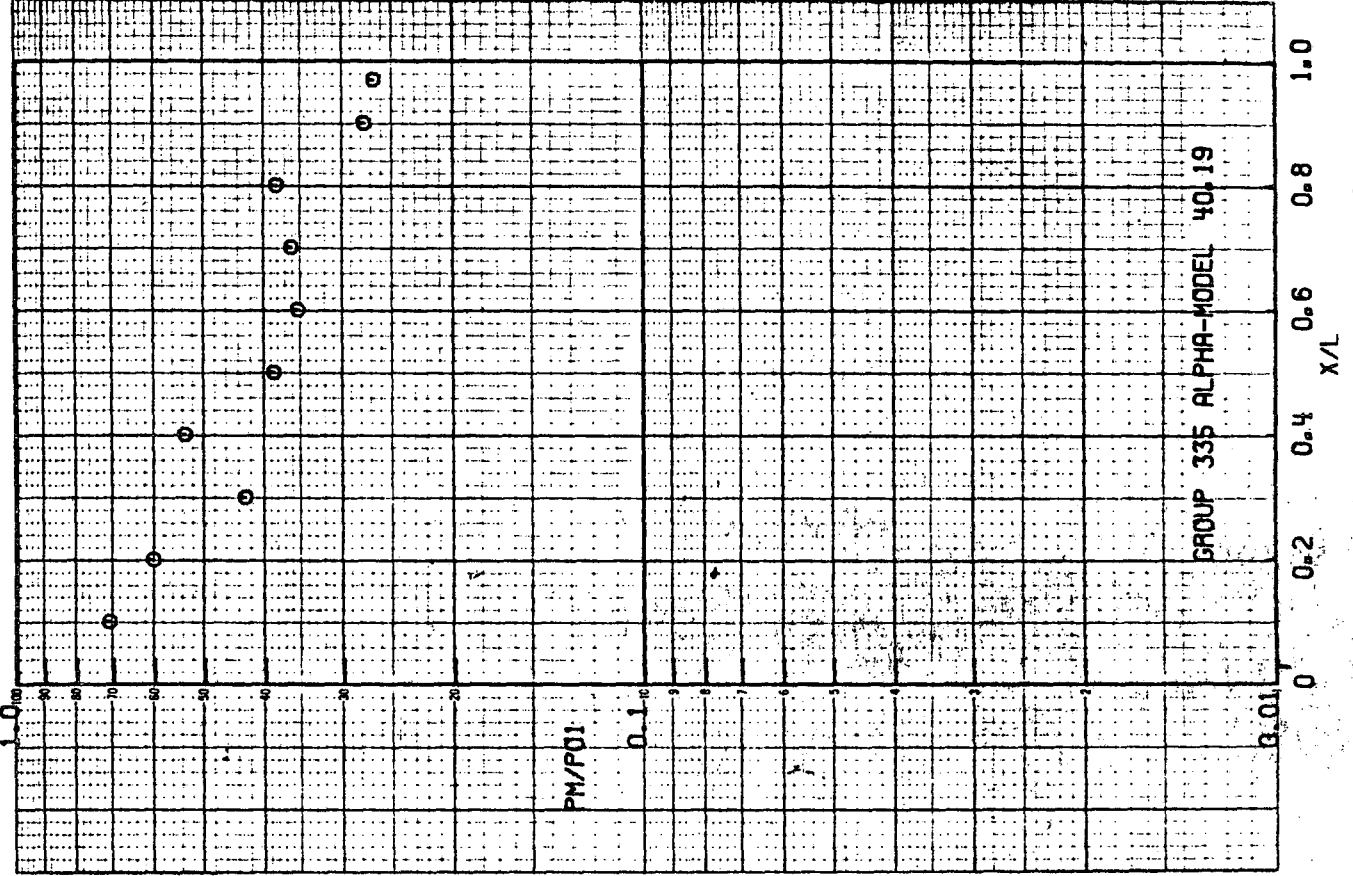
SURFACE PRESSURE AND FLOW FIELD DATA

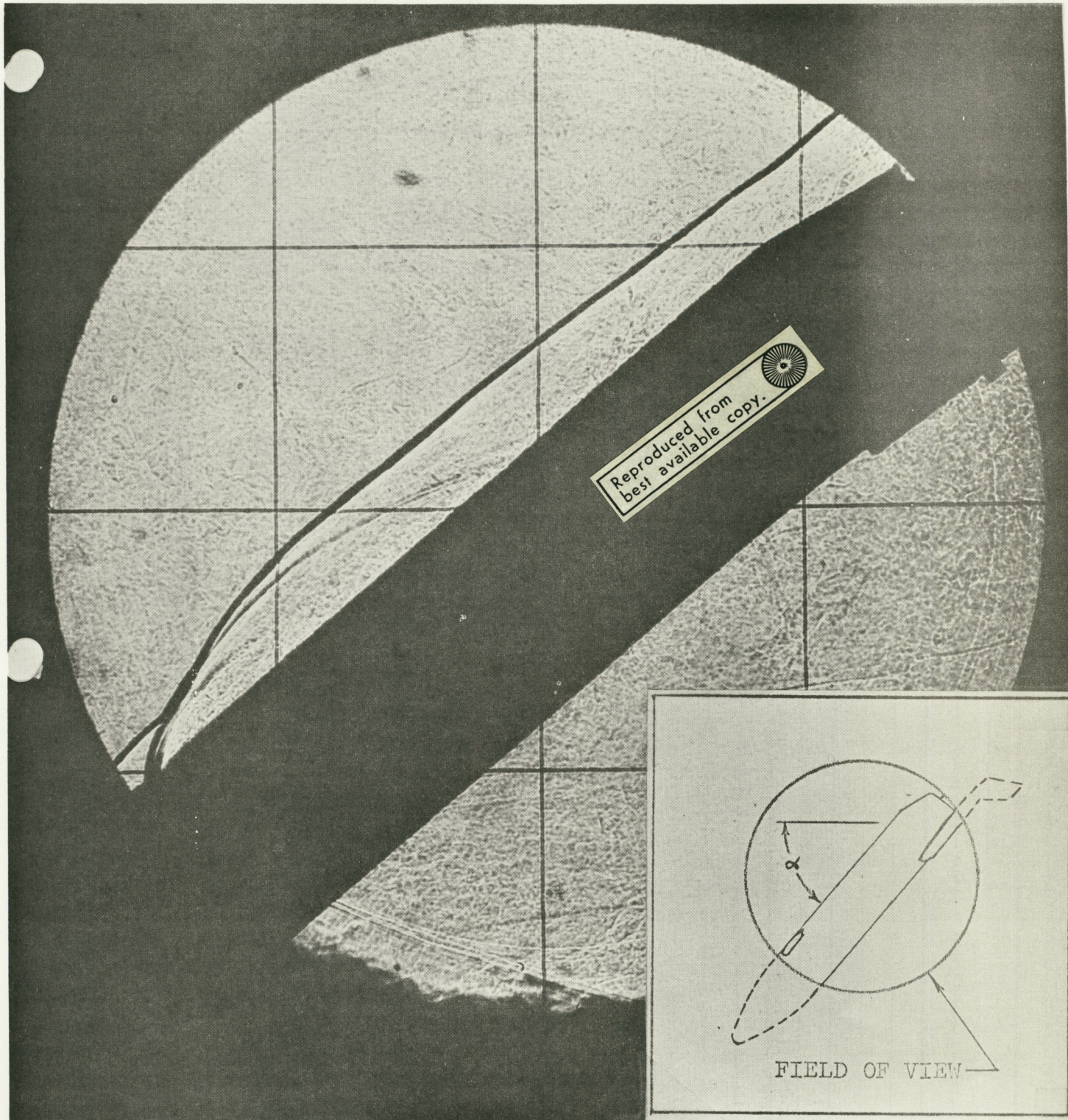


AEDC (ARO, INC.) ARNOLD AFS, TENNESSEE  
 VON KARMAN GAS DYNAMICS FACILITY  
 50 INCH HYPERSONIC TUNNEL R  
 VT1162

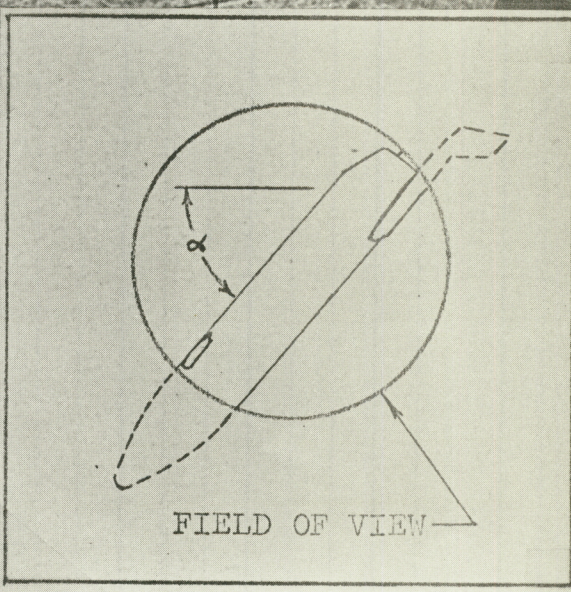
GROUP 335 CONFIG 42 MACH NO. 8.00 PO PSIA 860.8 TO DEG R 1344 ALPHA-MODEL 40.19 ALPHA-SECTOR 9.81 ALPHA-PREBEND -50.00 ROLL-MODEL 180.00 YAW .0

GROUP	CONFIG	MODEL	MACH NO.	PO PSIA	TO DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW
335	42	MDAC-B	8.00	860.8	1344	40.19	9.81	-50.00	180.00	.0
		P-INF (PSIA)	P01 (PSIA)	Q-INF (PSIA)	V-INF (FT/SEC)	RHO-INF (SLUGS/FT3)	MU-INF (LB-SEC/FT2)	RE/FT (FT-1)	L (IN)	
		8.82E-02	7.307	3.950	3870	2.444E-03	7.841E-08	3.750E 06	23.78	
CM	POS	TAP	PH	PM/PO	PH/P-01	PH/P-INF	CP	CP/CP-MAX	X/L	
			(PSIA)							
1	2	1	5.191E 00	6.030E-03	7.105E-01	5.887E 01	1.292E 00	7.069E-01	.100	
2	2	2	4.417E 00	5.131E-03	6.045E-01	5.010E 01	1.096E 00	5.997E-01	.200	
3	2	3	3.149E 00	3.657E-03	4.309E-01	3.571E 01	7.747E-01	4.240E-01	.300	
4	2	4	3.930E 00	4.566E-03	5.379E-01	4.457E 01	9.727E-01	5.323E-01	.400	
5	2	5	2.829E 00	3.286E-03	3.871E-01	3.208E 01	6.937E-01	3.796E-01	.500	
6	2	6	2.589E 00	3.007E-03	3.543E-01	2.936E 01	6.330E-01	3.464E-01	.600	
7	2	7	2.651E 00	3.079E-03	3.628E-01	3.006E 01	6.487E-01	3.550E-01	.700	
8	2	8	2.800E 00	3.253E-03	3.833E-01	3.176E 01	6.866E-01	3.757E-01	.800	
9	2	9	2.033E 00	2.361E-03	2.782E-01	2.305E 01	4.923E-01	2.694E-01	.900	
10	2	10	1.961E 00	2.278E-03	2.684E-01	2.224E 01	4.742E-01	2.595E-01	.970	





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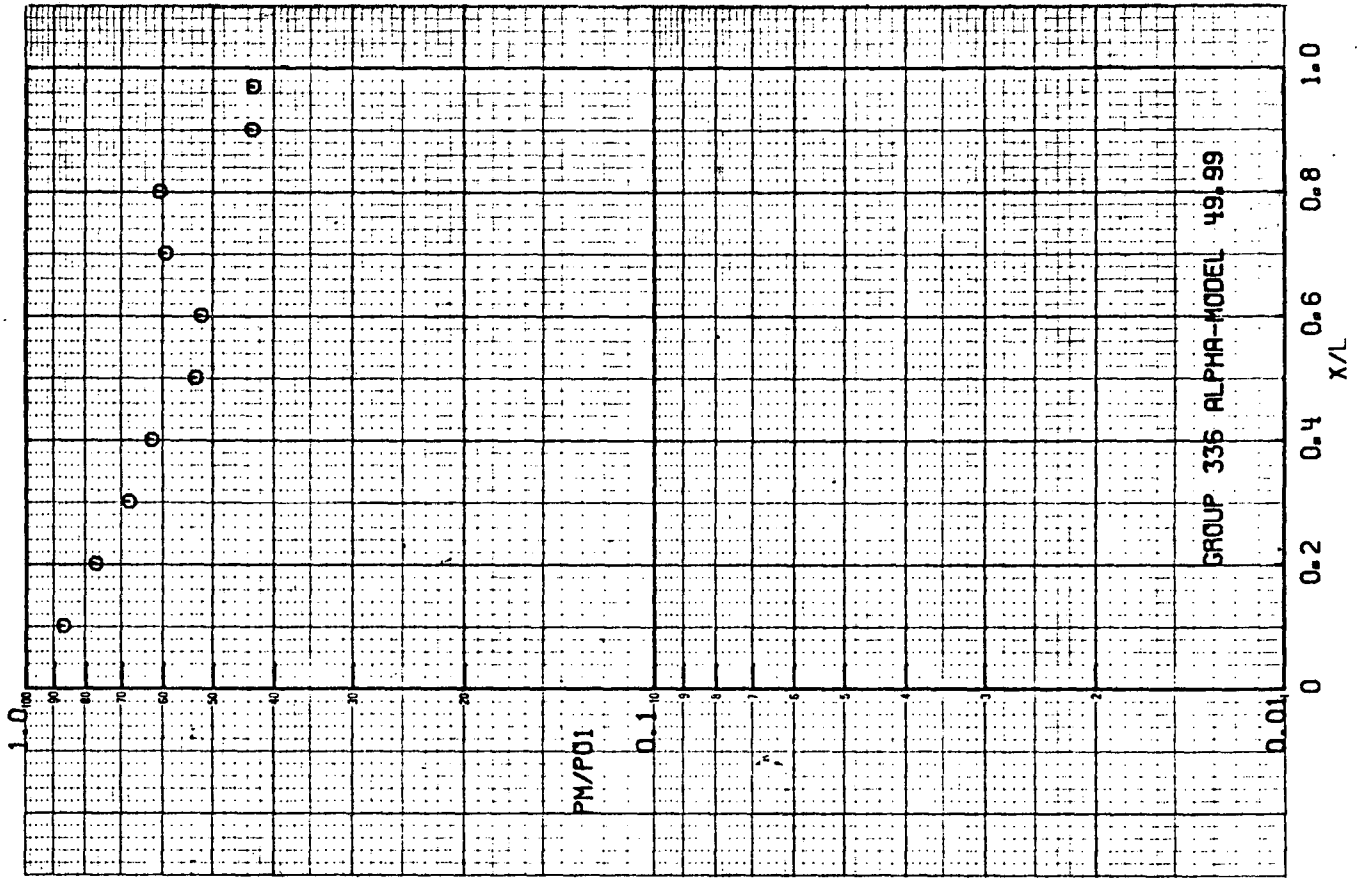
Group 335,  $\alpha = 40$

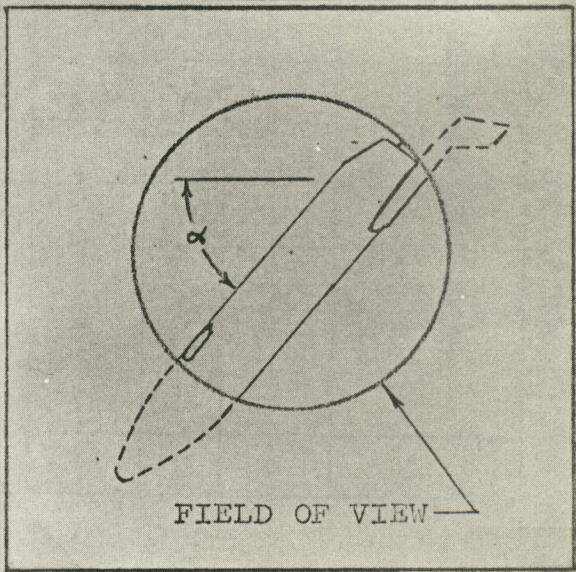
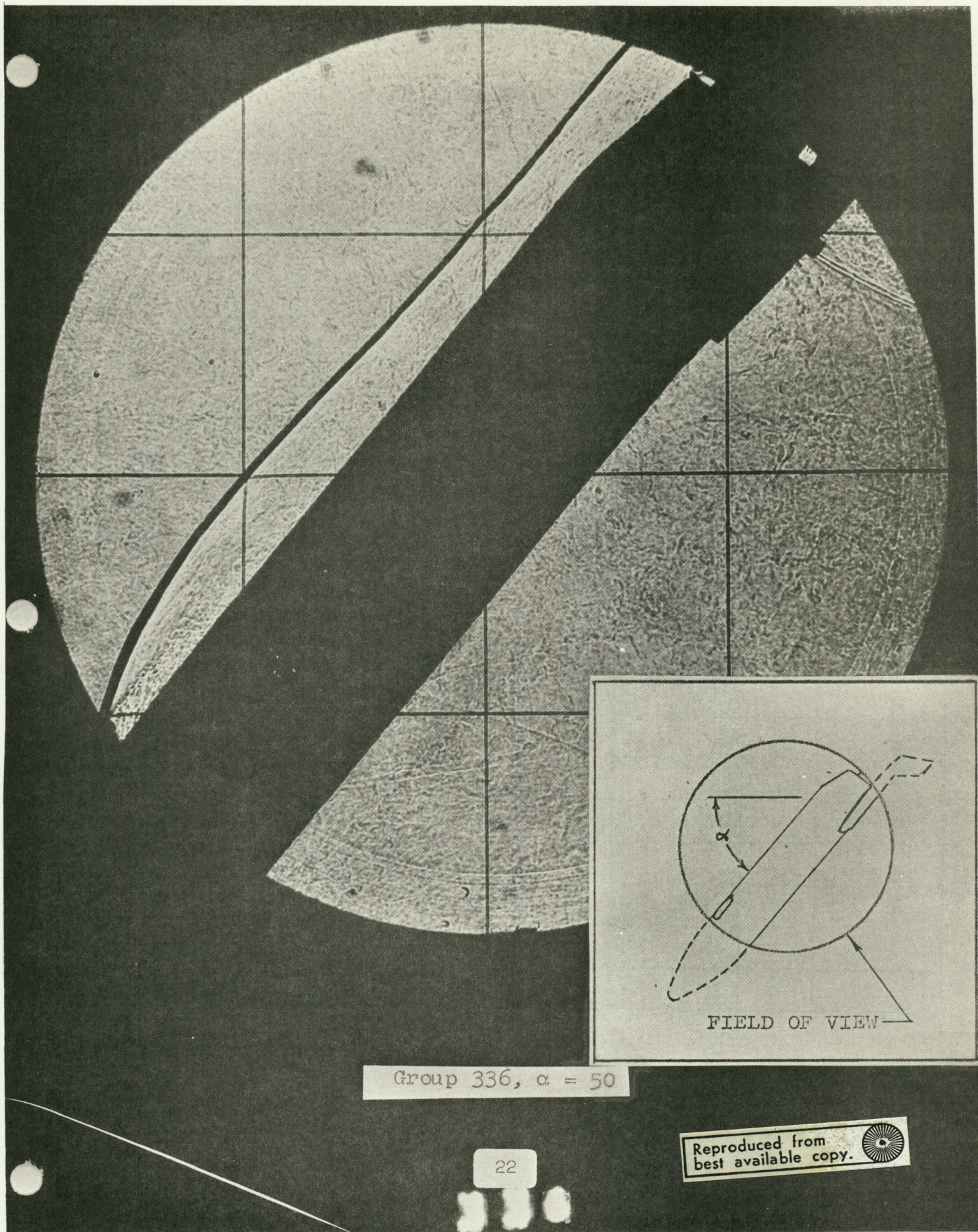
335

AEDC (ARO, INC.) ARNOLD AFS, TENNESSEE  
 VON KARMAN GAS DYNAMICS FACILITY  
 50 INCH HYPERSONIC TUNNEL B  
 V11162

GROUP 336 CONFIG 42 MODEL MDAC-B MACH NO. 8.00 PO PSIA 861.7 TO DEG R 1346 ALPHA-MODEL 50.00 ALPHA-SECTOR ALPHA-PREBEND ROLL-MODEL YAW  
 8.83E-02 7.314 3.954 3873 2.443E-03 7.853E-08 3.746E-06 23.78 100.00 -50.00

T-INF (DEG R)	P-INF (PSIA)	P01 (PSIA)	0-INF (PSIA)	V-INF (FT/SEC)	RHO-INF (SLUGS/FT3)	MU-INF (LB-SEC/FT2)	RE/FT (FT-1)	L (IN)
98	8.83E-02	7.314	3.954	3873	2.443E-03	7.853E-08	3.746E-06	23.78
CH	POS	TAP	PM (PSIA)	PM/PO	PM/PO1	CP	CP/CP-MAX	X/L
1	2	1	6.358E 00	7.378E-03	8.693E-01	7.203E 01	1.586E 00	8.677E-01
2	2	2	5.636E 00	6.540E-03	7.705E-01	6.385E 01	1.403E 00	7.677E-01
3	2	3	4.988E 00	5.788E-03	6.819E-01	5.950E 01	1.239E 00	6.780E-01
4	2	4	4.584E 00	5.320E-03	6.268E-01	5.194E 01	1.137E 00	6.222E-01
5	2	5	3.899E 00	4.524E-03	5.330E-01	4.417E 01	9.636E-01	5.273E-01
6	2	6	3.814E 00	4.426E-03	5.214E-01	4.321E 01	9.422E-01	5.156E-01
7	2	7	4.356E 00	5.055E-03	5.956E-01	4.935E 01	1.079E 00	5.906E-01
8	2	8	4.454E 00	5.168E-03	6.089E-01	5.046E 01	1.104E 00	6.041E-01
9	2	9	3.169E 00	3.678E-03	4.333E-01	3.590E 01	7.791E-01	4.264E-01
10	2	10	3.157E 00	3.663E-03	4.316E-01	3.576E 01	7.750E-01	4.246E-01





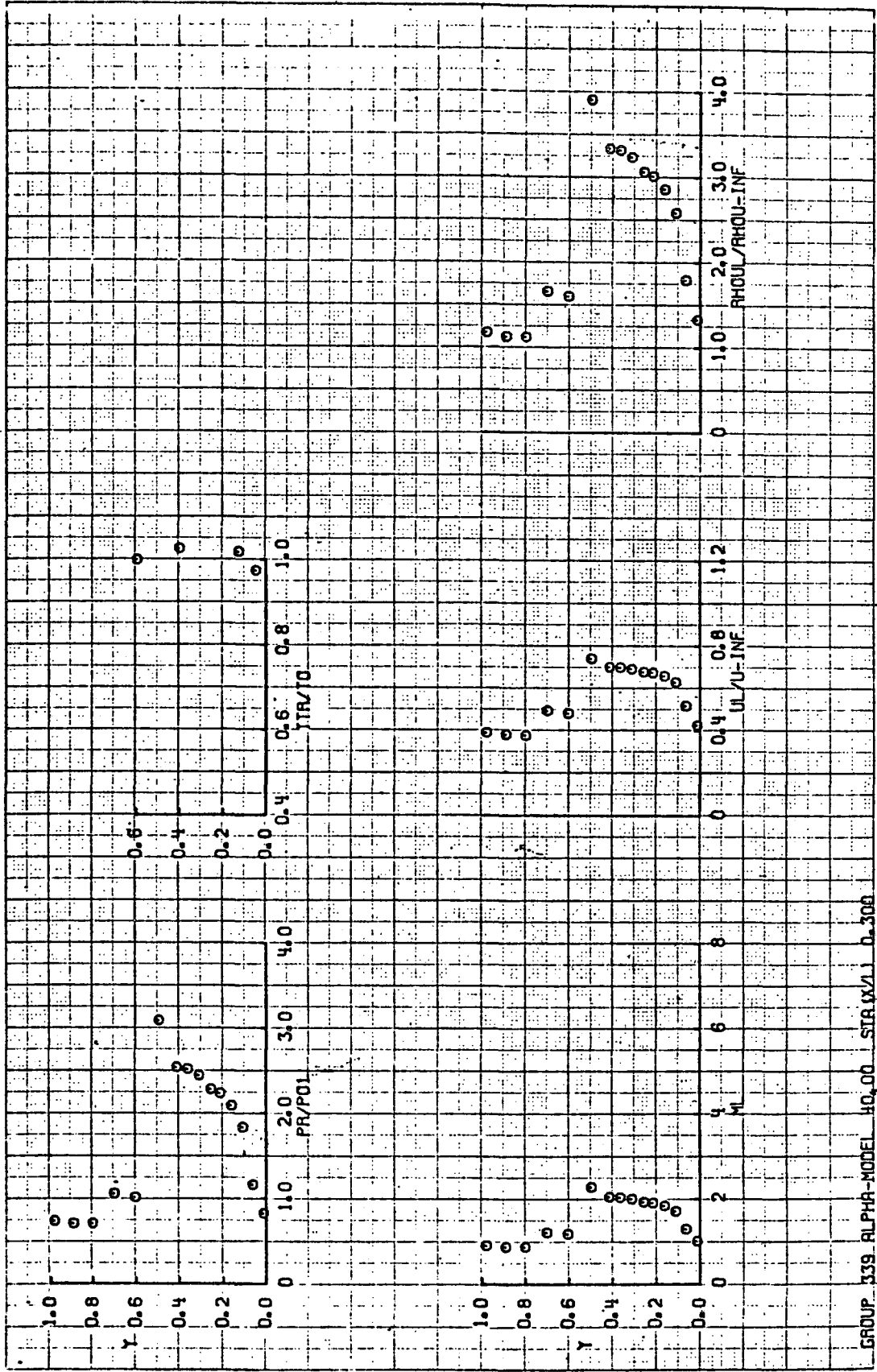
Group 336,  $\alpha = 50$



AEDC (ARG. INC.) ARNOLD AFS, TENNESSEE  
VON KARMAN GAS DYNAMICS FACILITY  
50 INCH HYPERSONIC TUNNEL B  
V11162

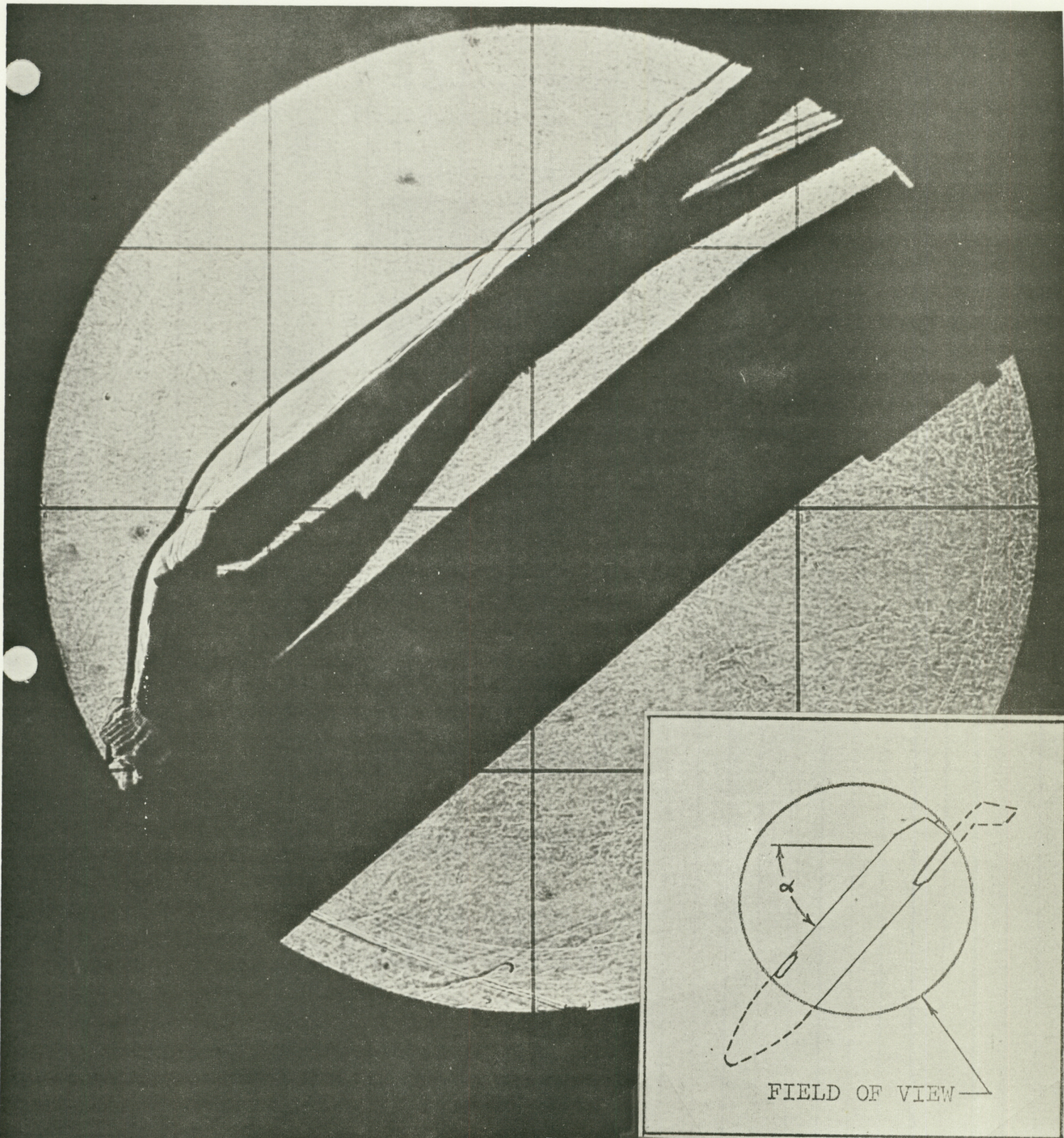
GROUP	CONFIG	MODEL	MACH NO.	PO PSIA	TO DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW			
239	42	MDAC-B	8.00	860.5	1343	40.01	9.99	-50.00	180.00	0			
I-Inf	(DEG R)	P-Inf	(PSIA)	U-Inf	(FT/SEC)	RHO-Inf	(LB-SEC/FT <sup>3</sup> )	RE/FT	MODEL STA	L			
97	8.81E-02	7.30A	3.949	3868	2.445E-03	7.835E-08	3.753E-06		300	23.78			
CM POS	TAP	PH	PR/PO1	Y(IN)	PML/PH	ML	REL	IL/T-Inf	UL/U-Inf	RHO/L/RHO-Inf	RHOUL/RHOUL-Inf	MUL/MU-Inf	
1	3	1	6.019E-00	8.241E-01	0.14	5.230E-01	1.009	6.076E-05	11.466	0.427	3.1147	1.3301	0.216
2	3	2	8.479E-00	1.141E-00	0.66	3.713E-01	1.294	8.786E-05	10.338	0.520	3.4546	1.7968	7.675
3	3	3	1.340E-01	1.433E-00	0.112	2.349E-01	1.708	1.418E-06	8.715	0.630	4.0980	2.5832	6.837
4	3	4	1.527E-01	2.091E-00	0.163	2.062E-01	1.839	1.634E-06	8.232	0.660	4.3382	2.8615	6.572
5	3	5	1.632E-01	2.236E-00	0.216	1.929E-01	1.909	1.761E-06	7.982	0.674	4.4746	3.0172	6.430
6	3	6	1.667E-01	2.292E-00	0.258	1.899E-01	1.931	1.801E-06	7.906	0.679	4.5173	3.0657	6.388
7	3	7	1.785E-01	2.444E-00	0.313	1.764E-01	2.005	1.946E-06	7.650	0.693	4.6685	3.2364	6.240
8	3	8	1.440E-01	2.520E-00	0.365	1.711E-01	2.040	2.018E-06	7.531	0.700	4.7421	3.3190	6.171
9	3	9	1.955E-01	2.540E-00	0.415	1.697E-01	2.048	2.035E-06	7.505	0.701	4.7586	3.3375	6.156
10	3	10	2.256E-01	3.099E-00	0.499	1.395E-01	2.278	2.567E-06	6.771	0.741	5.2747	3.9093	5.715
11	3	11	7.384E-01	1.011E-00	0.66	4.243E-01	1.179	7.608E-05	10.799	0.484	3.3071	1.6015	7.900
12	3	12	7.713E-00	1.058E-00	0.702	4.042E-01	1.214	7.955E-05	10.659	0.495	3.3507	1.6600	7.832
13	3	13	5.126E-00	7.059E-01	0.702	6.106E-01	0.870	4.980E-05	11.986	0.377	2.9798	1.1220	8.455
14	3	14	5.164E-00	7.077E-01	0.92	6.090E-01	0.872	4.999E-05	11.977	0.377	2.9820	1.1256	8.451
15	3	15	5.360E-00	7.339E-01	0.91	5.873E-01	0.906	5.254E-05	11.853	0.390	3.0131	1.1752	8.395

CM	TC	ITR	ITR/IC	Y(IN)	PML/PO1
1	1	1308	0.9799	0.51	4.310E-01
2	2	1367	1.0179	0.131	
5	5	1377	1.0253	0.402	
6	6	1341	0.9985	0.599	



GROUP 339 ALPHA-MODEL 40.00 STR XX/1 0.300



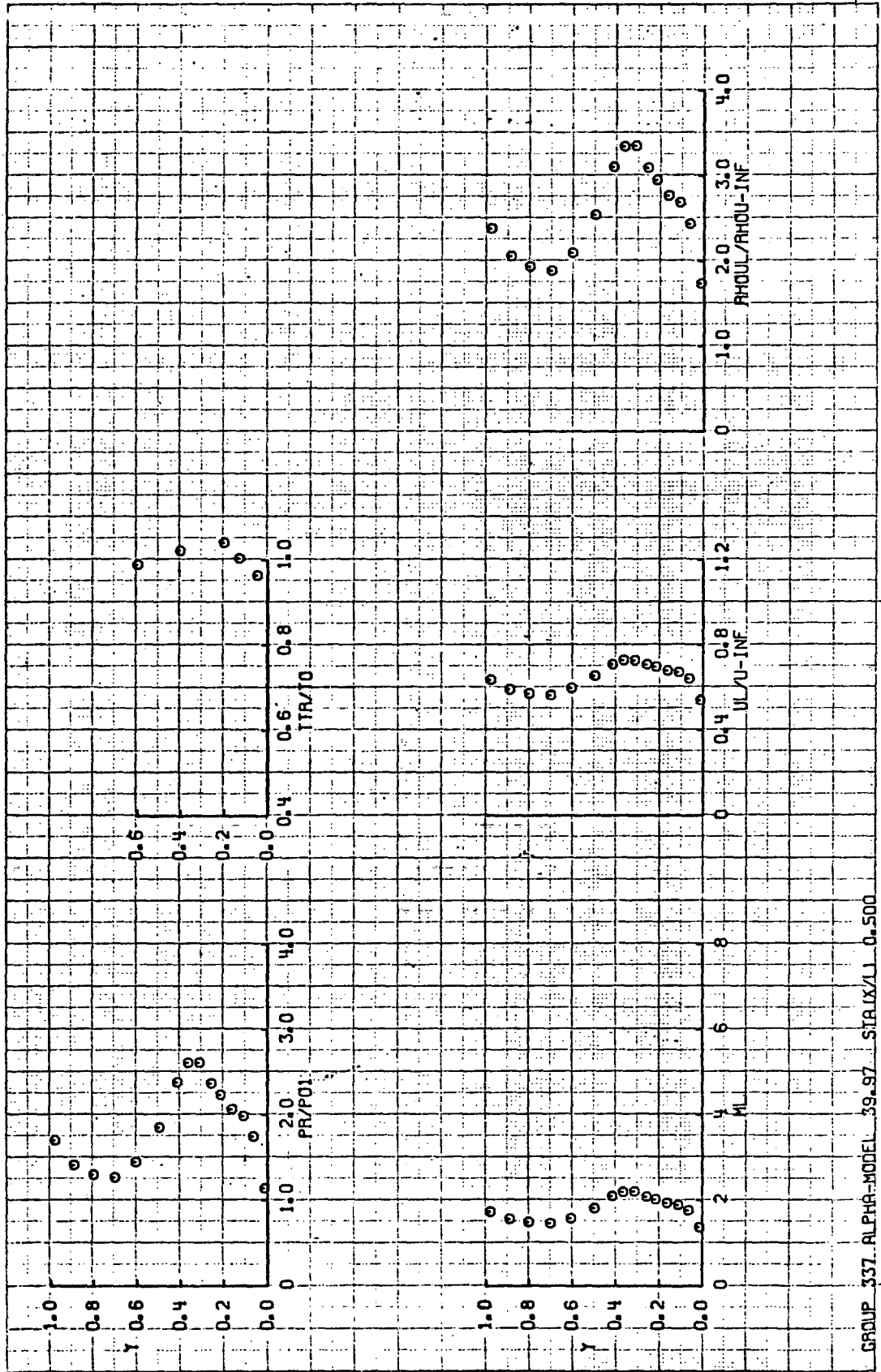


Group 339,  $\alpha = 40$ , RAKE STA (X/L) 0.3

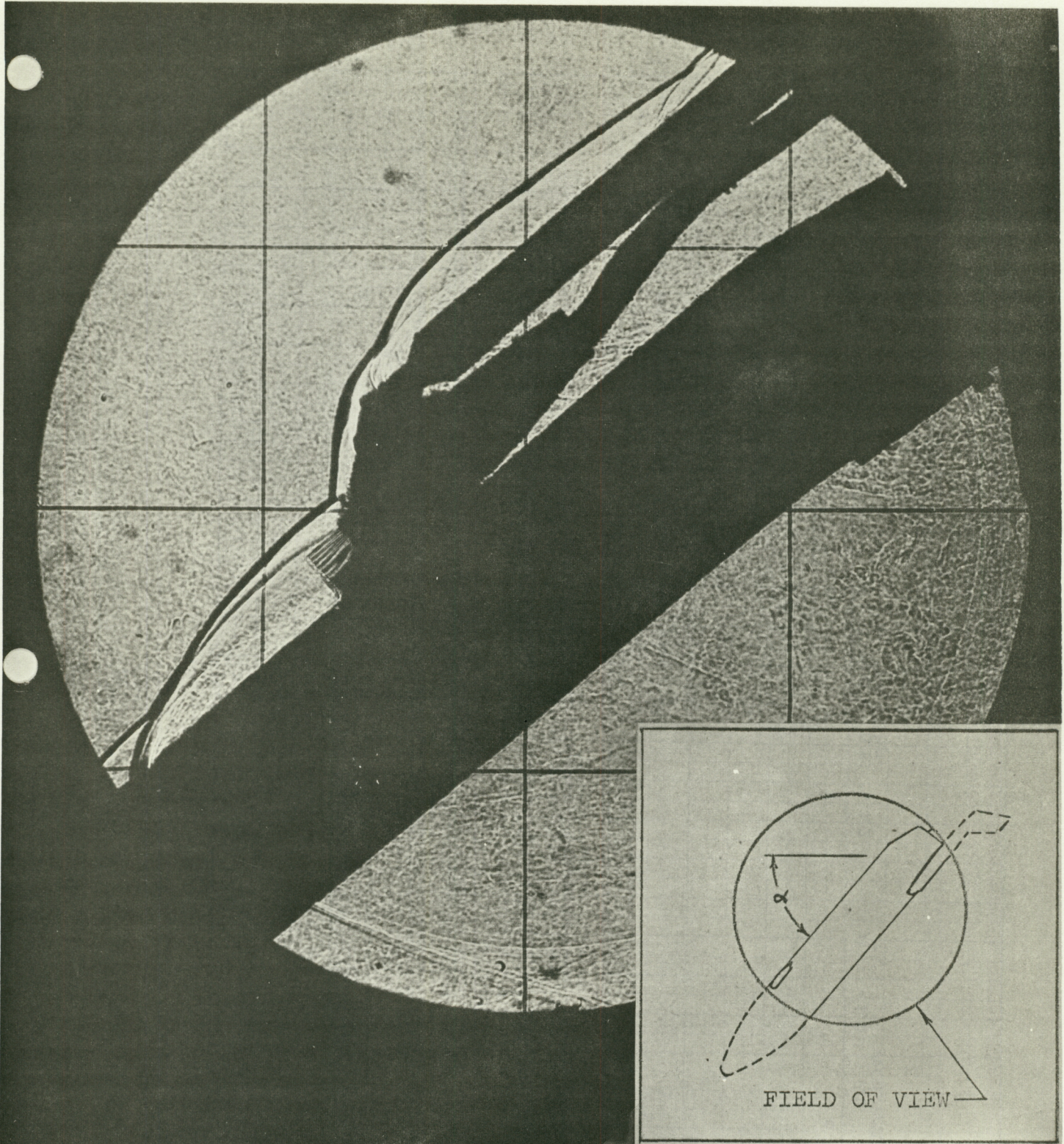
AEDC (ARO, INC.) ARNOLD AFS, TENNESSEE  
VON KARMAN GAS DYNAMICS FACILITY  
50 INCH HYPERSONIC TUNNEL R  
X11182

GROUP	CONFIG	MODEL	MACH NO.	PO (PSIA)	110 DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW			
337	42	MDAC-B	8.00	802.3	1344	39.97	10.03	-58.00	180.00	.0			
T-INF	(DEG H)	P-INF	(PSIA)	Q-INF	U-INF	RHO-INF	MU-INF	RE/FT	MODEL STA	L			
97	8.83E-02	7.319	3.957	3870	(FT/SEC)	(SLUGS/FT <sup>3</sup> )	(LB-SEC/FT <sup>2</sup> )	(FT-1)	(X/L)	(IN)			
CH	POS	TAP	PH	PR/PO1	Y(IN)	PML/PM	ML	REL	IL/1-INF	RHO/RHO-INF	RHOUL/RHOUL-INF	MUL/MU-INF	
									(FT-1)				
1	3	1	8.358E-00	1.142E-00	.14	3.389E-01	1.372	8.682E-05	10.025	.543	3.1987	1.7373	7.517
2	3	2	1.281E-01	1.751E-00	.06	2.211E-01	1.769	1.362E-06	8.409	.644	3.7775	2.4334	6.712
3	3	3	1.455E-01	1.987E-00	.112	1.947E-01	1.899	1.567E-06	8.016	.672	4.0005	2.6895	6.449
4	3	4	1.514E-01	2.066E-00	.163	1.871E-01	1.940	1.636E-06	7.872	.681	4.0737	2.7726	6.367
5	3	5	1.636E-01	2.235E-00	.216	1.732E-01	2.026	1.789E-06	7.577	.697	4.2322	2.9512	6.197
6	3	6	1.734E-01	2.369E-00	.258	1.636E-01	2.093	1.915E-06	7.356	.710	4.3593	3.0934	6.067
7	3	7	1.915E-01	2.616E-00	.313	1.480E-01	2.208	2.152E-06	6.987	.730	4.5896	3.3488	5.846
8	3	8	1.912E-01	2.612E-00	.365	1.482E-01	2.206	2.148E-06	6.993	.729	4.5856	3.3444	5.850
9	3	9	1.746E-01	2.429E-00	.215	1.624E-01	2.099	1.927E-06	7.337	.711	4.3707	3.1061	6.056
10	3	10	1.355E-01	1.851E-00	.099	2.090E-01	1.825	1.448E-06	8.282	.657	3.8721	2.5426	6.598
11	3	11	1.059E-01	1.447E-00	.506	2.675E-01	1.583	1.109E-06	9.193	.600	3.4884	2.0831	7.028
12	3	12	9.266E-00	1.266E-00	.702	3.057E-01	1.462	9.652E-05	9.658	.568	3.3171	1.8849	7.336
13	3	13	9.534E-00	1.205E-00	.802	2.965E-01	1.489	9.964E-05	9.560	.576	3.3546	1.9310	7.281
14	3	14	1.037E-01	1.417E-00	.92	2.731E-01	1.563	1.045E-06	9.269	.595	3.4599	2.0588	7.130
15	3	15	1.248E-01	1.705E-00	.891	2.269E-01	1.741	1.322E-06	8.591	.638	3.7329	2.3816	6.788

CH	TC	TTR	TTR/TC	Y(IN)	PML/PO1
1	1	1294	.4628	.051	3.870E-01
2	2	1349	1.0037	.131	
3	3	1398	1.0442	.202	
5	5	1373	1.0216	.402	
6	6	1329	.9888	.599	



GROUP 337 ALPHA-MODEL 39.97 STA IX/1 0.500



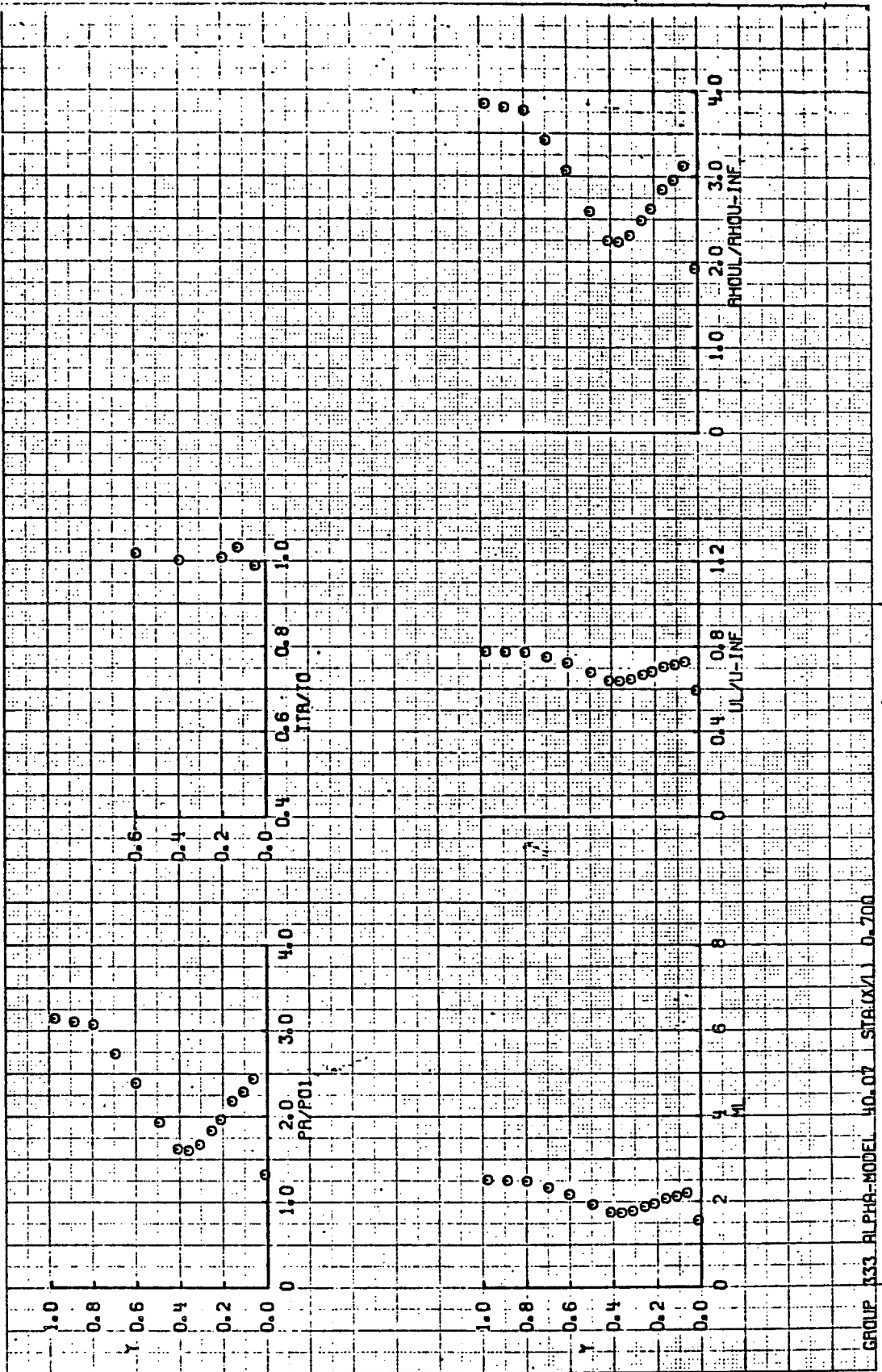
Group 337,  $\alpha = 40$ , RAKE STA (X/L) 0.5

337

AEDC (AKO, INC.) ARNOLD AFS, TENNESSEE  
 VON KARMAN GAS DYNAMICS FACILITY  
 50 INCH HYPERSONIC TUNNEL R  
 VJ1162

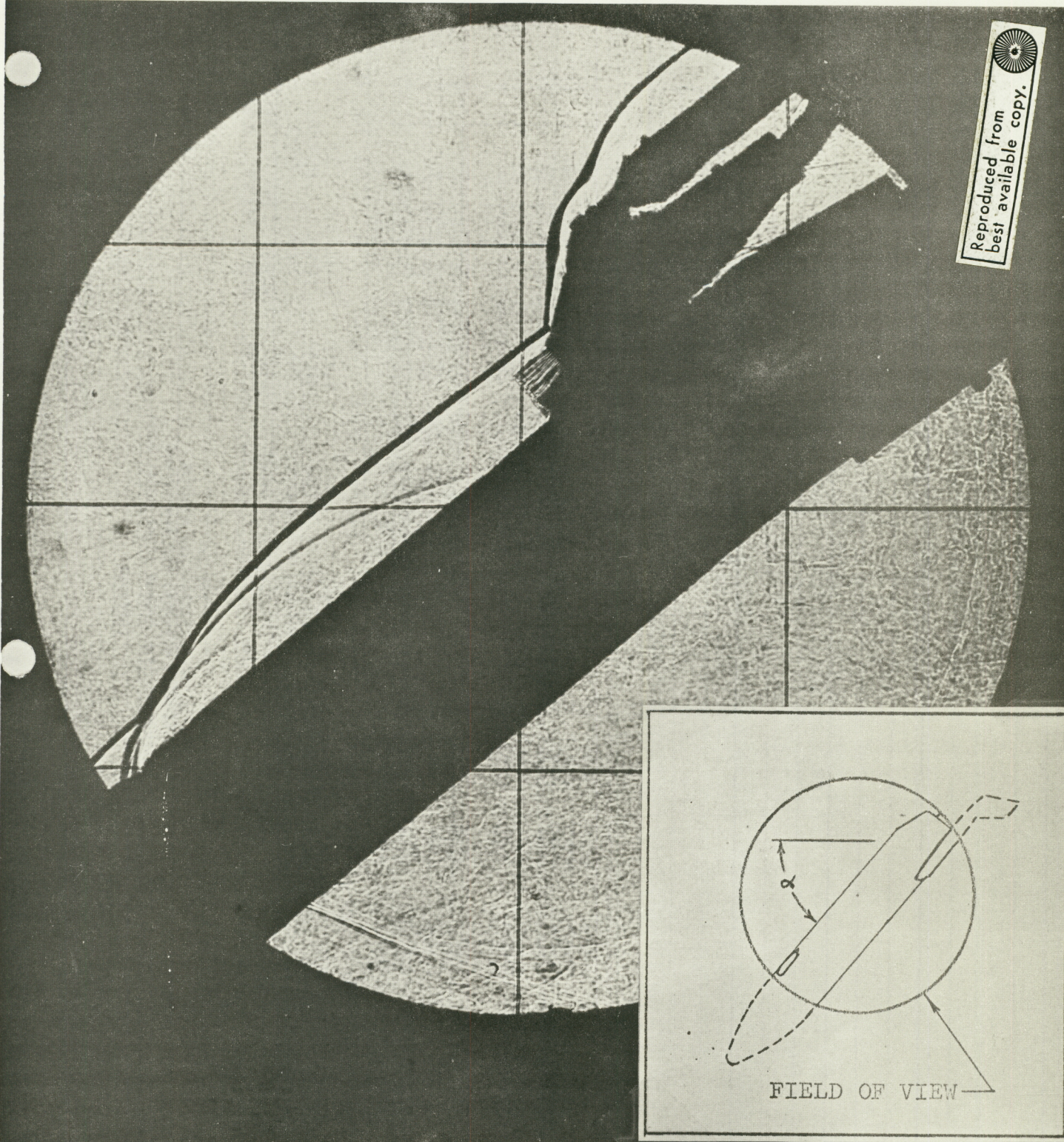
GROUP	CONFIG	MODEL	MACH NO.	PO	PSIA	10 DEG R	ALPHA-MODEL	ALPHA-SECTION	ALPHA-PREBEND	ROLL-MODEL	YAW		
333	42	MDAC-R	8.00	861.0	1342	40.07			-50.00	180.00	.0		
T-INF	P-INF	POI	Q-INF	U-INF	RHO-INF	MU-INF	RE/FT	MODEL STA	L				
(DEG R)	(PSIA)	(PSIA)	(PSIA)	(FT/SEC)	(SLUGS/FT3)	(LB-SEC/FT2)	(FT-1)	(X/L)	(IN)				
97	8.42E-02	7.30A	3.951	3867	2.449E-03	7.829E-08	3.759E-06	.100	23.78				
CH	POS	IAP	PH	PR/POI	Y(IN)	PML/PH	ML	REL	TL/T-INF	UL/U-INF	RHOL/RHO-INF	RHOUL/RHOU-INF	MUL/MU-INF
								(FT-1)					
1	2	1	9.689E 00	1.225E 00	.014	2.740E-01	1.562	1.016E 06	9.276	.595	3.242E	1.9279	7.137
2	3	2	1.784E 01	2.441E 00	.66	1.487E-01	2.202	2.007E 06	7.005	.729	4.2937	3.1287	5.859
3	3	3	1.671E 01	2.246E 00	.112	1.589E-01	2.124	1.955E 06	7.254	.715	4.1464	2.9655	6.009
4	3	4	1.592E 01	2.174E 00	.163	1.660E-01	2.069	1.755E 06	7.434	.705	4.0464	2.8541	6.115
5	3	5	1.439E 01	1.964E 00	.216	1.849E-01	1.954	1.597E 06	7.826	.683	3.8443	2.6270	6.242
6	3	6	1.340E 01	1.834E 00	.258	1.979E-01	1.882	1.443E 06	8.078	.669	3.7235	2.4897	6.487
7	3	7	1.221E 01	1.671E 00	.313	2.173E-01	1.786	1.302E 06	8.425	.648	3.5704	2.3140	6.680
8	3	8	1.170E 01	1.601E 00	.365	2.267E-01	1.743	1.243E 06	8.584	.638	3.5043	2.2374	6.767
9	3	9	1.143E 01	1.519E 00	.415	2.243E-01	1.753	1.256E 06	8.547	.641	3.5192	2.2547	6.747
10	3	10	1.412E 01	1.931E 00	.499	1.879E-01	1.937	1.529E 06	7.886	.680	3.8145	2.5932	6.377
11	3	11	1.749E 01	2.392E 00	.506	1.514E-01	2.177	1.357E 06	7.085	.724	4.2453	3.0751	5.908
12	3	12	1.997E 01	2.733E 00	.702	1.374E-01	2.339	2.299E 06	6.590	.751	4.5644	3.4260	5.603
13	3	13	2.240E 01	3.074E 00	.902	1.181E-01	2.491	2.664E 06	6.157	.773	4.8852	3.7753	5.328
14	3	14	2.270E 01	3.106E 00	.92	1.169E-01	2.505	2.699E 06	6.120	.775	4.9149	3.8075	5.303
15	3	15	2.300E 01	3.147E 00	.981	1.153E-01	2.522	2.745E 06	6.072	.777	4.9535	3.8492	5.272

CH	TC	TR	TH/TC	Y(IN)	PML/POI
		(DEG R)	(DEG R)		
1	1	1327	.988	.051	3.630E-01
2	2	1394	1.2317	.131	
3	3	1452	1.40075	.202	
5	5	1344	1.0015	.402	
6	6	1366	1.0175	.599	



GROUP 333 ALPHA-MODEL 40.07 STR(X)/1 0-700

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Group 333,  $\alpha = 40$ , RAKE STA (X/L) 0.7

333

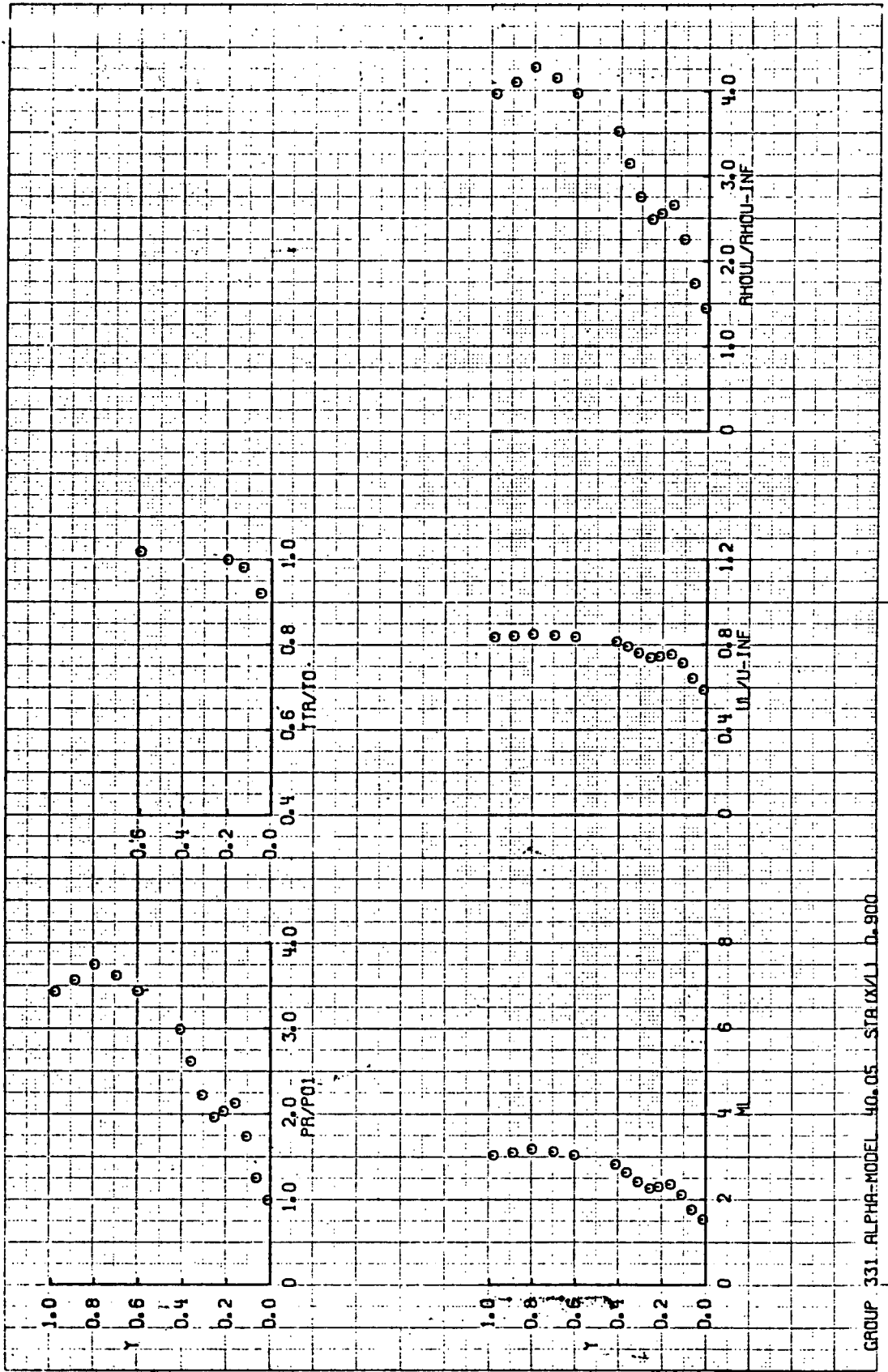
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 PEDC (ARO, INC.) ARNOLD AFS, TENNESSEE  
 VON KARMAN GAS DYNAMICS FACILITY  
 50 INCH HYPERSONIC TUNNEL #  
 VI1162

GROUP	CURFIG	MODEL	MACH NO.	PO	PSIA	TO DEG R	ALPHA-MODEL	ALPHA-SECTION	ALPHA-PREBEND	ROLL-MODEL	YAW	
331	42	MDAC-H	8.00	85H-9	1342	40.05	9.95	-50.00	180.00		.0	
T-INF	(DEG R)	P-INF	(PSIA)	POI	O-INF	U-INF	RHO-INF	MU-INF	RE/FT	MODEL STA	L	
97		H.79E-02	7.283	3.237	3M67	(FT/SEC)	(SLUGS/FT3)	(LB-SEC/FT2)	(FT-1)	(IN/L)	(IN)	
CH	POS	TAP	PR/POI	Y(IN)	PML/PR	ML	REF	TL/T-INF	UL/U-INF	RHOL/RHO-INF	MUL/MU-INF	
							(FT-1)					
1	3	1	7.20E-01	9.892E-01	.114	2.810E-01	1.538	7.546E-05	9.349	2.4591	1.4472	7.185
2	3	2	9.112E-01	1.251E-01	.066	2.222E-01	1.763	9.890E-05	9.511	2.7066	1.7400	6.727
3	3	3	1.268E-01	1.716E-01	.112	1.602E-01	2.114	1.402E-06	7.206	3.1617	2.2557	6.028
4	3	4	1.548E-01	2.127E-01	.163	1.307E-01	2.360	1.791E-06	6.527	3.5293	2.6606	5.564
5	3	5	1.478E-01	2.031E-01	.215	1.369E-01	2.302	1.691E-06	6.700	3.4381	2.5609	5.672
6	3	6	1.423E-01	1.963E-01	.258	1.416E-01	2.261	1.624E-06	6.824	3.3756	2.4923	5.749
7	3	7	1.617E-01	2.220E-01	.313	1.252E-01	2.415	1.889E-06	6.370	2.6165	2.7557	5.664
8	3	8	1.403E-01	2.613E-01	.265	1.064E-01	2.632	2.322E-06	5.785	3.9818	3.1511	5.083
9	3	9	2.177E-01	2.390E-01	.415	9.298E-02	2.925	2.174E-06	5.315	4.3340	3.5291	4.762
11	3	11	2.508E-01	3.638E-01	.700	8.047E-02	3.040	3.366E-06	4.845	4.7547	3.9775	4.427
12	3	12	2.536E-01	3.619E-01	.702	7.682E-02	3.122	3.617E-06	4.679	4.5235	4.1567	4.305
13	3	13	2.730E-01	3.749E-01	.702	7.416E-02	3.179	3.799E-06	4.568	5.0427	4.2830	4.223
14	3	14	2.594E-01	3.567E-01	.692	7.794E-02	3.099	3.544E-06	4.726	4.8748	4.1050	4.340
15	3	15	2.498E-01	3.430E-01	.681	8.104E-02	3.036	3.354E-06	4.853	4.7468	3.9691	4.433

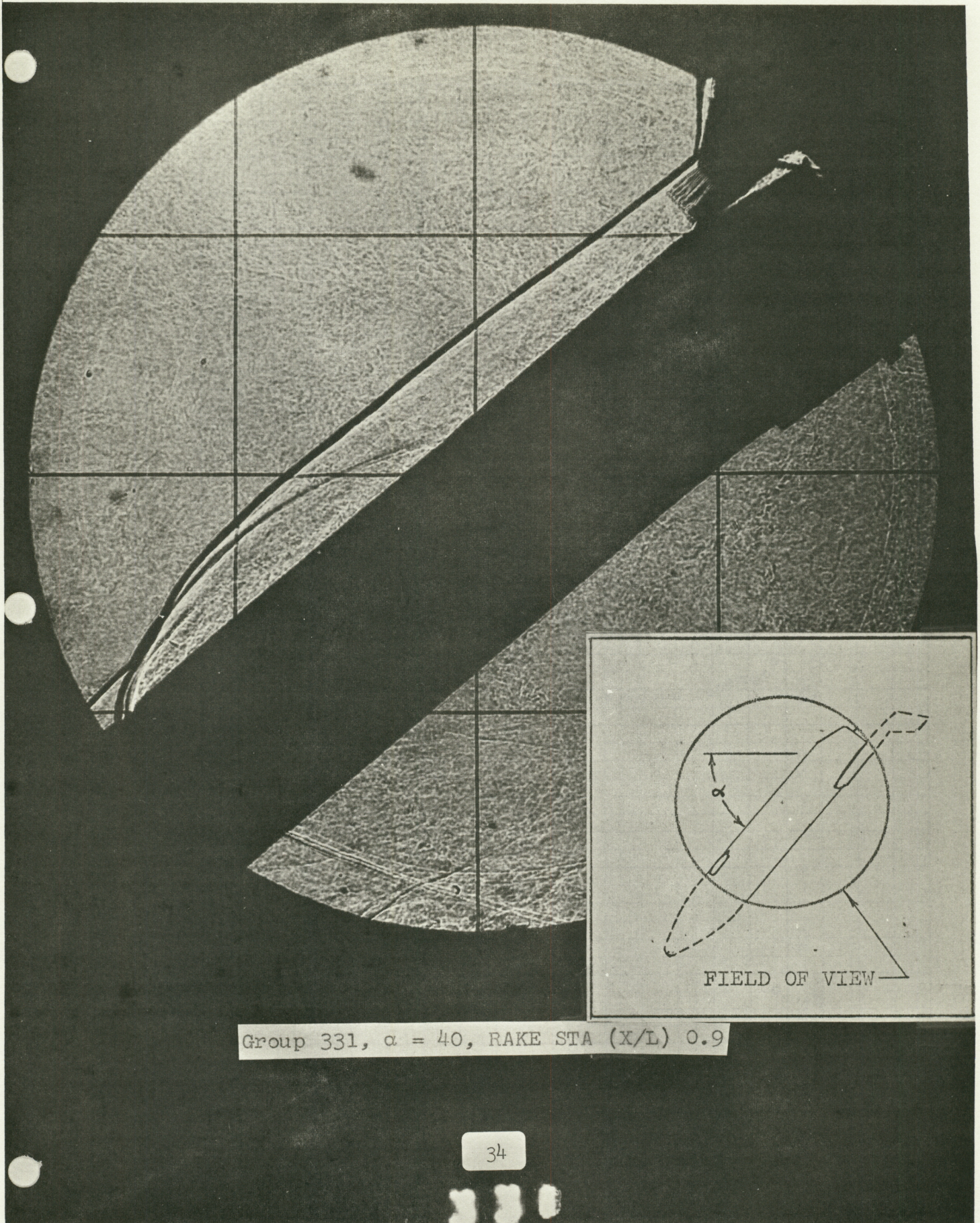
CH	IC	IR	IR/IC	Y(IN)	PML/POI
		(DEG R)	(DEG R)		
1	1	1235	.9203	.051	2.780E-01
2	2	1317	.9814	.131	
3	3	1341	.9992	.202	
6	6	1366	1.0179	.599	

GROUP  
331





GROUP 331 ALPHA-MODEL 40.05 STA.XV11 D.900

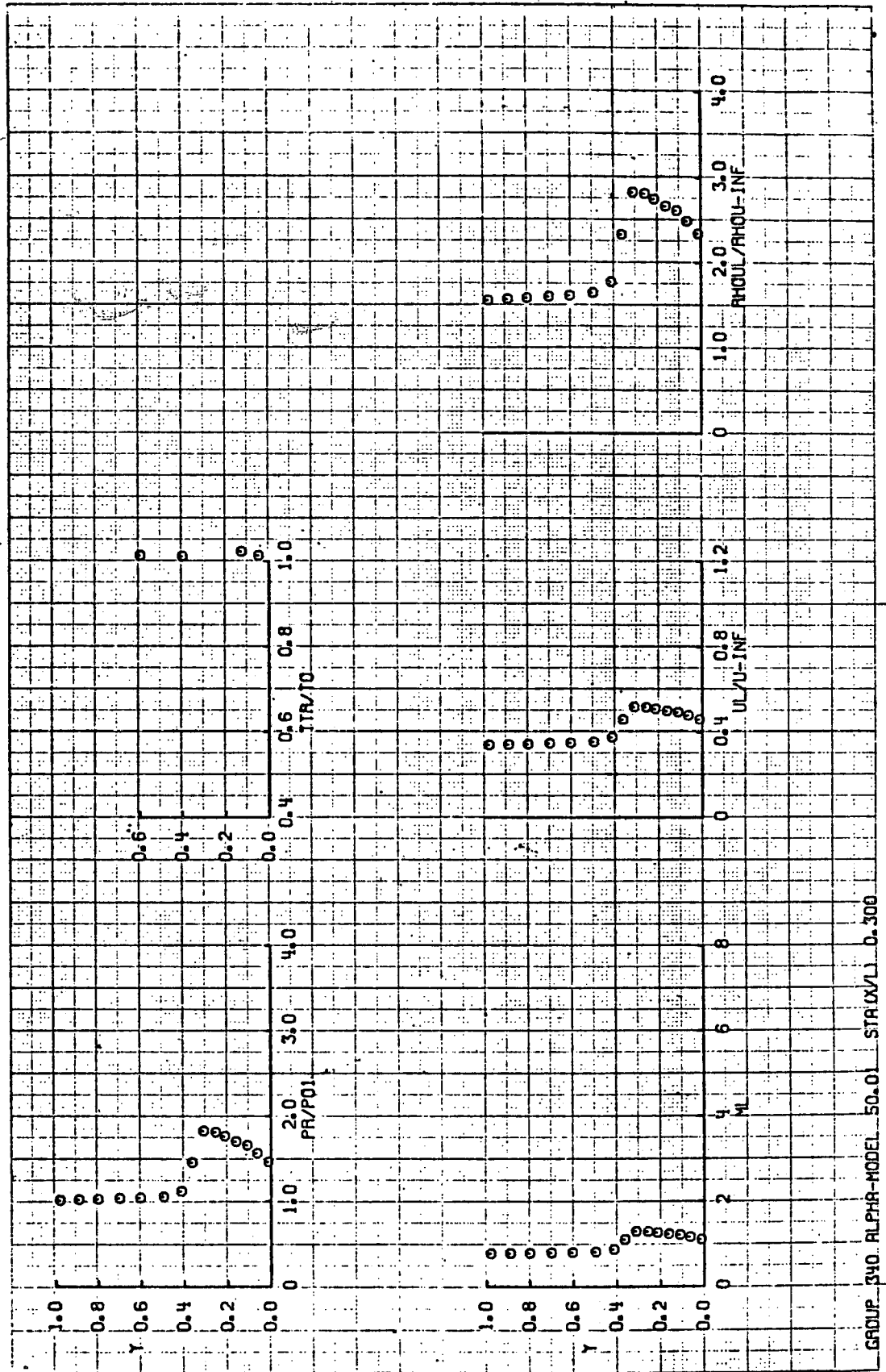


Group 331,  $\alpha = 40$ , RAKE STA (X/L) 0.9

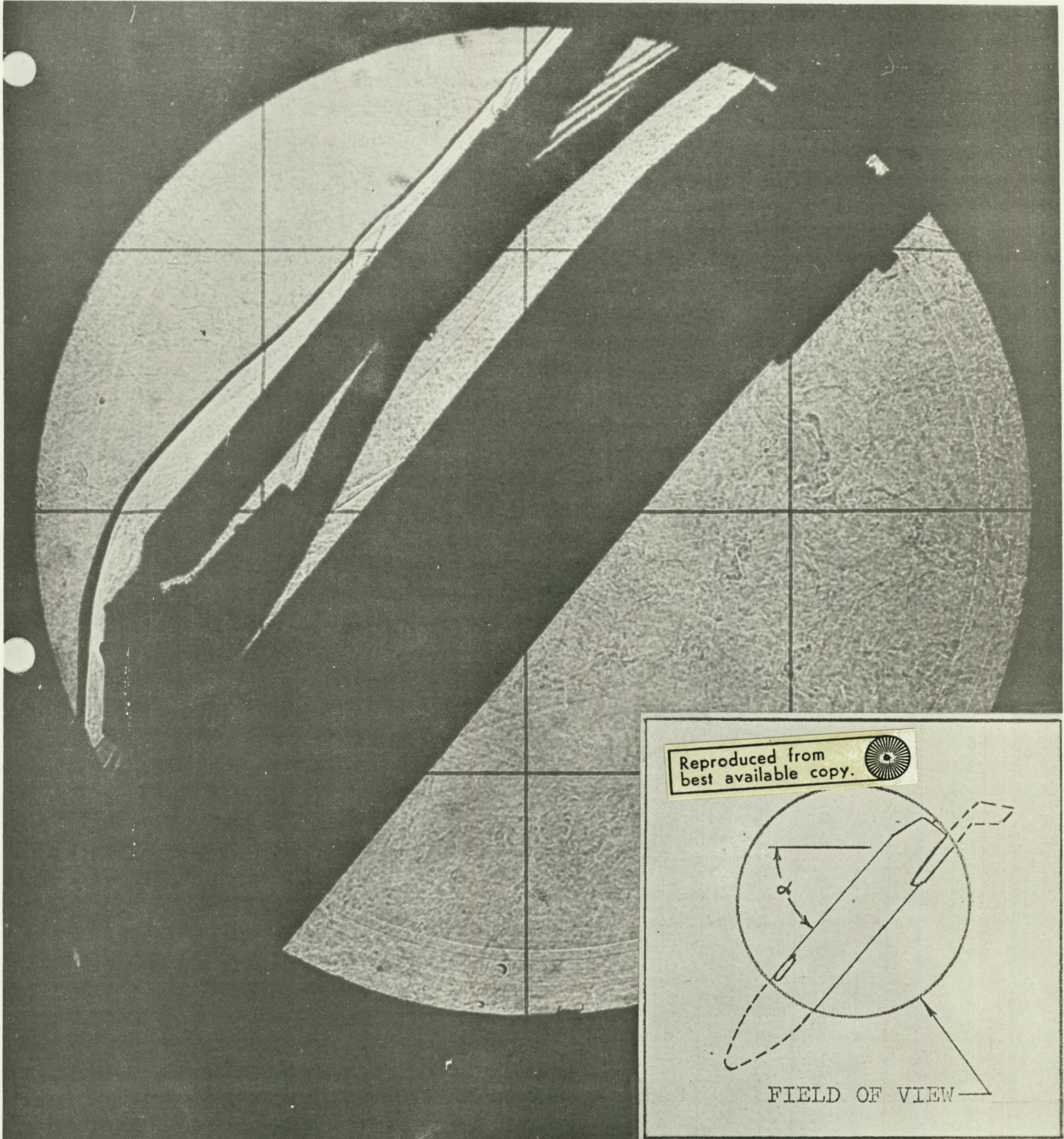
EDC (ARO, INC.) ARNOLD AFS, TENNESSEE  
VON KARMAN GAS DYNAMICS FACILITY  
50 INCH HYPERSONIC TUNNEL R  
V11162

GROUP	CONFIG	MODEL	MACH NO.	PO PSTA	TO DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW			
340	42	MDAC-B	8.00	850-2	134	50.01	-0.01	-50.00	180.00	0.0			
T-INF	(DEG R)	P-INF	(PSTIA)	POI	0-INF	U-INF	RHO-INF	MU-INF	RE/FT	MODEL STA			
97	8.81E-02	7.302	3.947	3870	(SLUGS/FT3)	(FT-1)	(LB-SEC/FT2)	(IN)	(IN)	23.78			
CH	POS	TAP	PR	PR/P01	Y(IN)	PML/PH	ML	REF	TL/T-INF	UL/U-INF	RH0L/RHO-INF	RHOUL/RHOU-INF	MUL/MU-INF
1	3	1	1.00E-01	1.462E-00	.214	4.064E-01	1.104	1.092E-06	11.093	.460	5.0943	2.3428	9.038
2	3	2	1.043E-01	1.564E-00	.266	4.356E-01	1.161	1.176E-06	10.869	.479	5.1994	2.4883	7.931
3	3	3	1.211E-01	1.658E-00	.312	4.115E-01	1.208	1.243E-06	10.682	.494	5.2904	2.6112	7.841
4	3	4	1.241E-01	1.699E-00	.363	4.014E-01	1.228	1.279E-06	10.604	.500	5.3293	2.6632	7.803
5	3	5	1.287E-01	1.763E-00	.416	3.864E-01	1.259	1.330E-06	10.479	.509	5.3930	2.7473	7.742
6	3	6	1.323E-01	1.812E-00	.458	3.743E-01	1.282	1.369E-06	10.395	.517	5.4417	2.8111	7.696
7	3	7	1.333E-01	1.825E-00	.493	3.736E-01	1.298	1.379E-06	10.362	.518	5.4541	2.8271	7.684
8	3	8	1.002E-01	1.455E-00	.365	4.647E-01	1.101	1.087E-06	11.109	.459	5.0872	2.3329	8.046
9	3	9	8.154E-00	1.117E-00	.415	6.104E-01	.870	7.874E-05	11.985	.377	4.7155	1.7760	8.453
10	3	10	7.684E-00	1.053E-00	.499	6.477E-01	.813	7.213E-05	12.190	.355	4.6362	1.6447	8.545
11	3	11	7.585E-00	1.039E-00	.500	6.566E-01	.799	7.061E-05	12.237	.349	4.6183	1.6141	8.567
12	3	12	7.545E-00	1.033E-00	.702	6.600E-01	.794	7.002E-05	12.255	.347	4.6114	1.6022	8.575
13	3	13	7.644E-00	1.025E-00	.502	6.645E-01	.786	6.911E-05	12.284	.344	4.6007	1.5838	8.588
14	3	14	7.645E-00	1.020E-00	.592	6.649E-01	.780	6.851E-05	12.302	.342	4.5937	1.5716	8.596
15	3	15	7.394E-00	1.013E-00	.581	6.734E-01	.773	6.775E-05	12.326	.339	4.5849	1.5561	8.607

CH	YC	TTR	TTW/TC	Y(IN)	PML/POI
1	1	1362	1.0134	.051	6.820E-01
2	2	1374	1.0223	.131	
5	5	1359	1.0112	.402	
6	6	1362	1.0134	.599	



GROUP 340 ALPHA-MODEL 50.01 STRA(X)/I 0.300



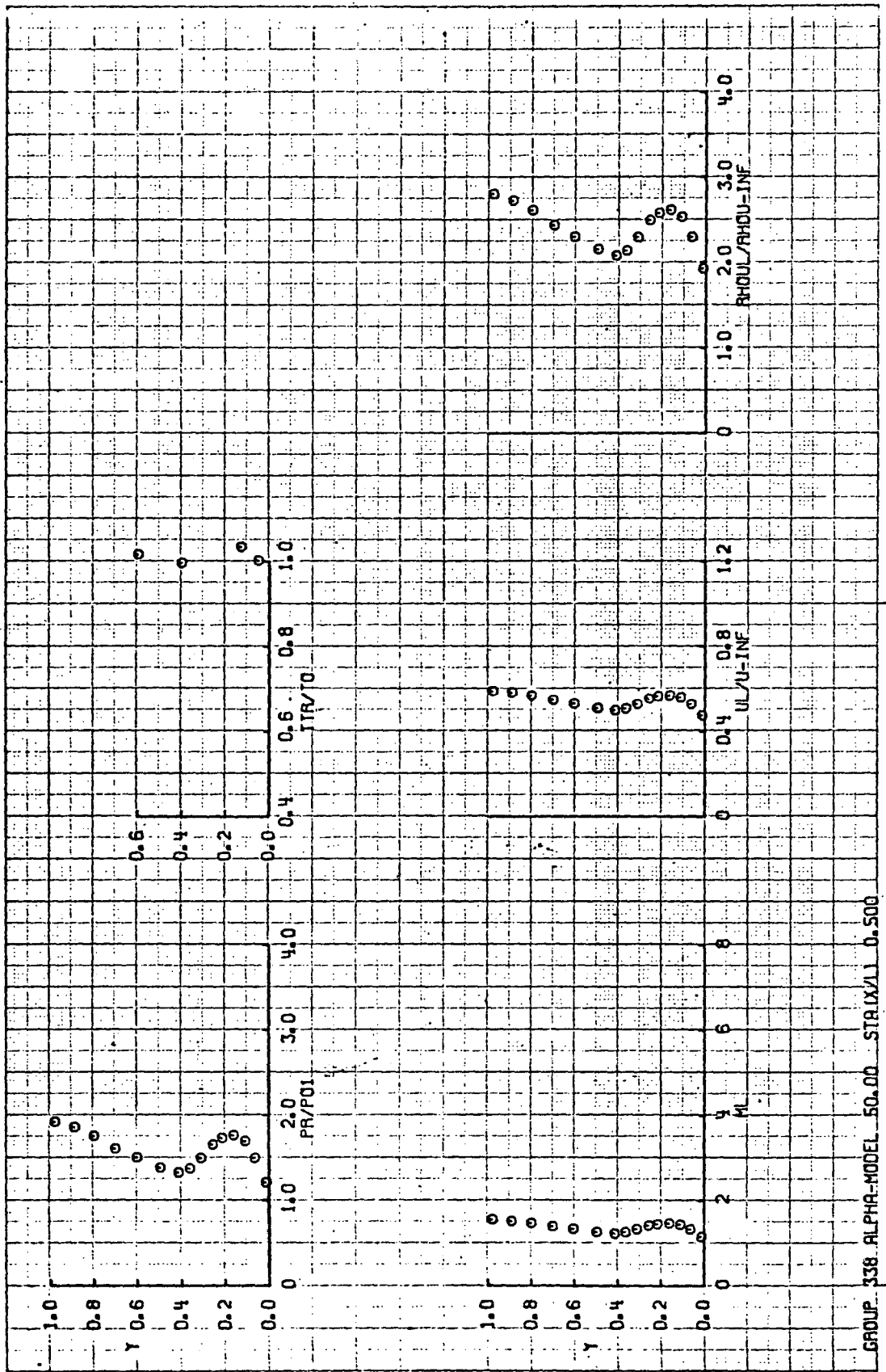
Group 340,  $\alpha = 50$ , RAKE STA (X/L) 0.3

EDC (ARO, INC.) ARNOLD AFS, TENNESSEE  
 VON KARMAN GAS DYNAMICS FACILITY  
 50 INCH HYPERSONIC TUNNEL B  
 VIIIA2

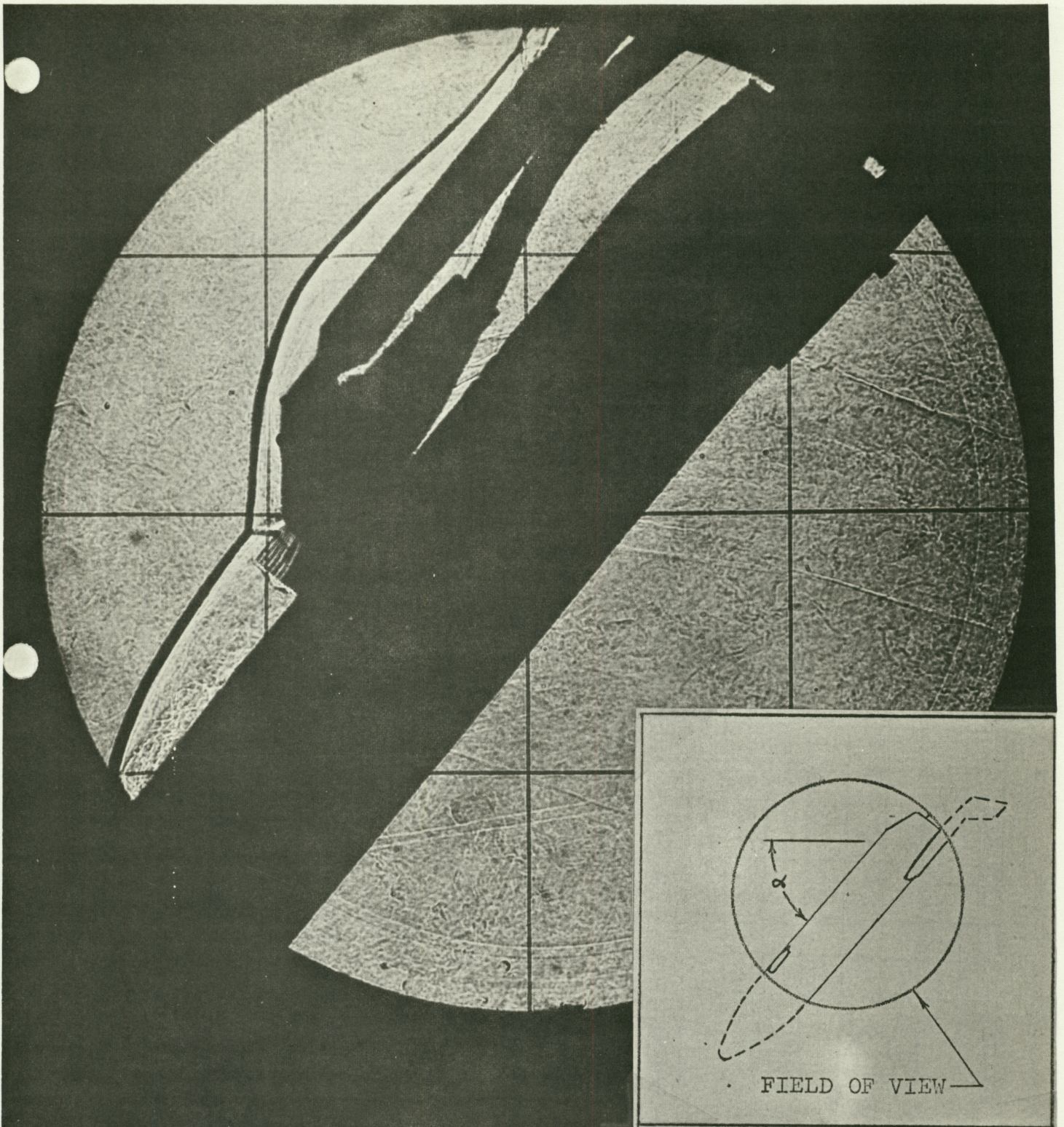
GROUP	CURTIS	MODEL	MACH NO.	PO	PSIA	TO DEG H	ALPHA-MONEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW		
33A	42	MDAC-H	8.00	862.6	1343	50.00	0	0	-50.00	180.00	0		
Y-1NF	(DEG R)	P-1NF	(PSIA)	P01	(PSIA)	Q-1NF	U-1NF	RHO-1NF	(SLUGS/FT3)	MU-1NF	RE/FT	MODEL STA	L
97	8.84E-02	7.322	3.95H	386H	2.451E-03	7.835E-08	3.162E-06	.500	23.78				
CH	POS	IAP	PR	PR/PO1	Y(IN)	PML/PR	ML	HFL	TL/T-1NF	UL/U-1NF	RHO/RHO-1NF	RHOUL/RHO-1NF	MUL/MU-1NF
1	2	1	8.871E-00	1.212E-00	.014	4.309E-01	1.153	9.130E-05	10.900	.476	4.0519	1.9288	7.948
2	3	2	1.094E-01	1.497E-00	.066	3.594E-01	1.329	1.137E-06	10.197	.531	4.3312	2.2981	7.605
3	3	3	1.241E-01	1.695E-00	.112	3.164E-01	1.437	1.292E-06	9.768	.561	4.5214	2.5378	7.389
4	3	4	1.291E-01	1.763E-00	.163	3.072E-01	1.472	1.346E-06	9.629	.571	4.5868	2.6186	7.318
5	3	5	1.249E-01	1.733E-00	.218	3.075E-01	1.456	1.322E-06	9.601	.567	4.5575	2.5826	7.450
6	3	6	1.213E-01	1.694E-00	.258	3.214E-01	1.417	1.263E-06	9.846	.556	4.4857	2.4934	7.429
7	3	7	1.097E-01	1.498E-00	.313	3.594E-01	1.329	1.137E-06	10.197	.531	4.3312	2.2981	7.605
8	3	8	1.006E-01	1.375E-00	.365	3.877E-01	1.257	1.041E-06	10.497	.509	4.2116	2.1429	7.748
9	3	9	9.705E-00	1.325E-00	.415	4.021E-01	1.228	1.003E-06	10.604	.500	4.1650	2.0814	7.805
10	3	10	1.013E-01	1.344E-00	.449	3.842E-01	1.263	1.044E-06	10.463	.511	4.2210	2.1553	7.736
11	3	11	1.102E-01	1.505E-00	.502	3.542E-01	1.333	1.152E-06	10.182	.532	4.3319	2.3066	7.597
12	3	12	1.180E-01	1.612E-00	.570	3.306E-01	1.394	1.238E-06	9.940	.549	4.4435	2.4406	7.476
13	3	13	1.284E-01	1.759E-00	.602	3.030E-01	1.470	1.343E-06	9.637	.570	4.5831	2.6141	7.322
14	3	14	1.362E-01	1.841E-00	.692	2.865E-01	1.521	1.424E-06	9.437	.584	4.6803	2.7330	7.219
15	3	15	1.411E-01	1.927E-00	.801	2.764E-01	1.552	1.476E-06	9.314	.592	4.7418	2.8074	7.155

CH	TC	TTR	TTR/TC	Y(IN)	PML/PO1
1	1	1346	1.0022	.051	5.330E-01
2	2	1349	1.0343	.131	
5	5	1339	.9970	.402	
6	6	1366	1.0171	.599	

GROUP  
33A



GROUP 338 ALPHA-MODEL 50.00 STRIX/11 0.500



Group 338,  $\alpha = 50$ , RAKE STA (X/L) 0.5

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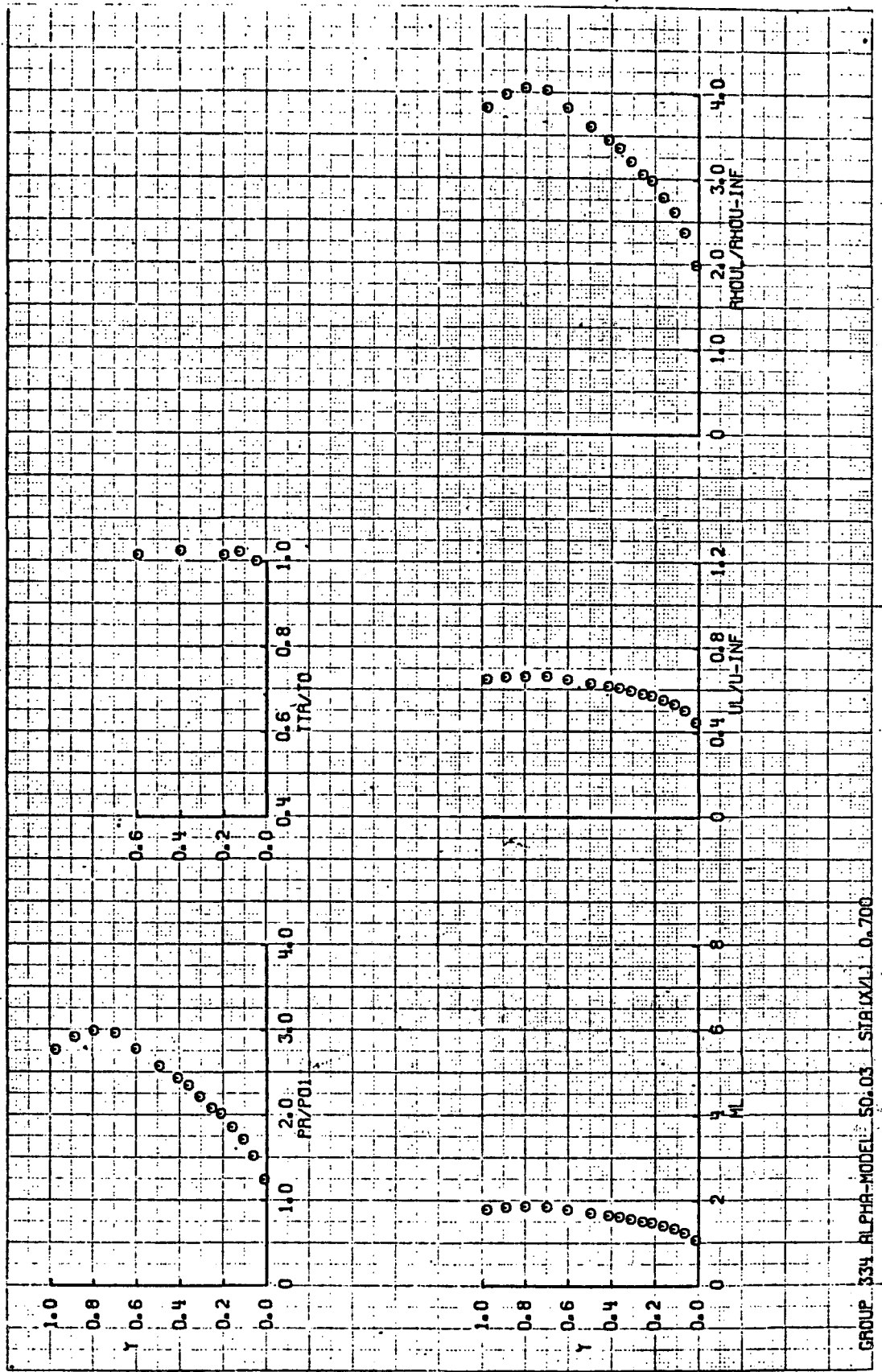
AEDC (ARO, INC.) ARNOLD AFS, TENNESSEE  
 VON KARMAN GAS DYNAMICS FACILITY  
 50 INCH HYPERSONIC TUNNEL R  
 VII162

GROUP 334 CONFIG 42 MACH NO. 8.00 MDEL M013 TO DEG R 1391 ALPHA-MODEL 50.03 ALPHA-SECTOR -03 ALPHA-PREBEND -50.00 ROLL-MODEL 180.00 YAW 0.0

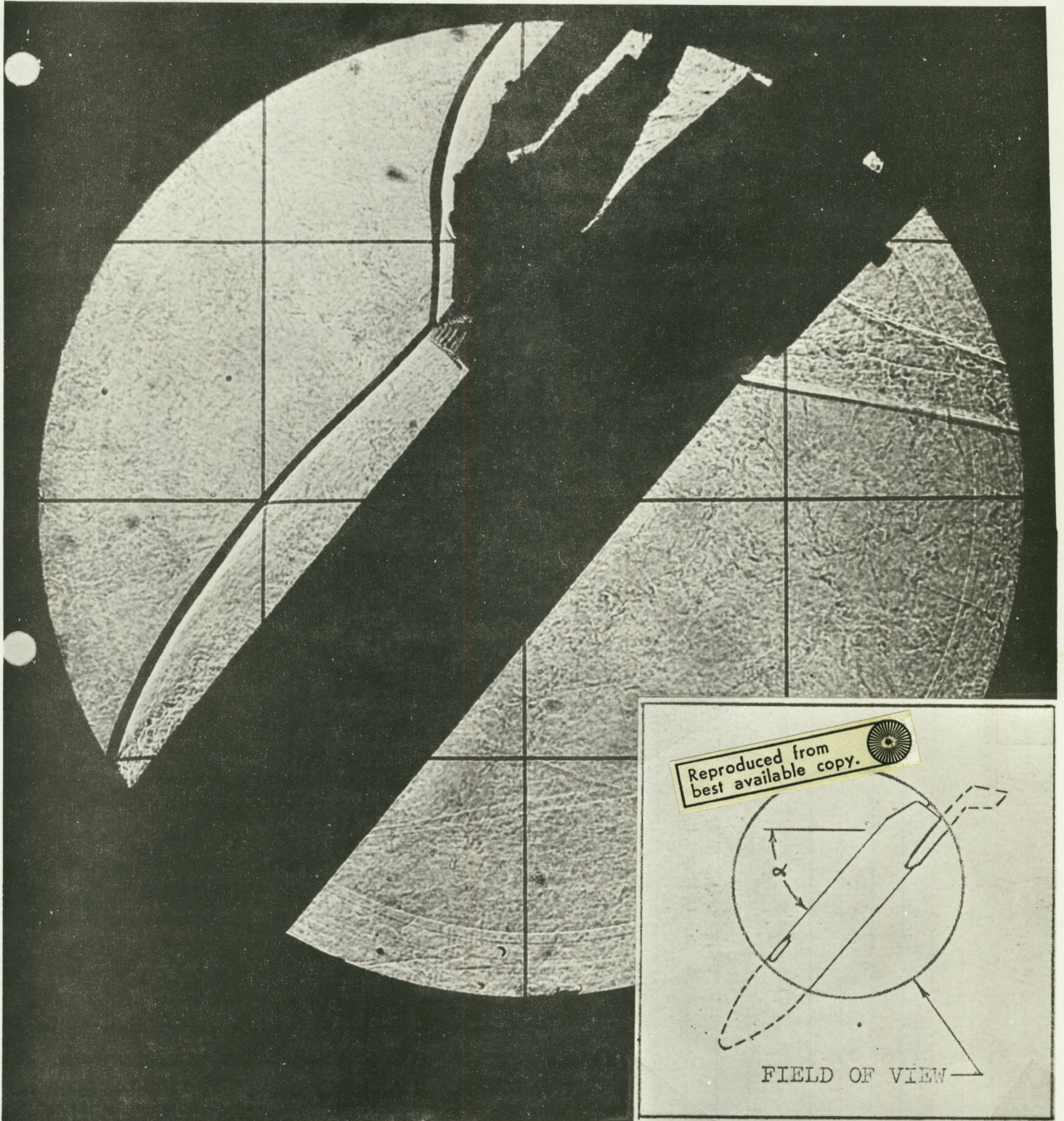
T-INF (DEG R) 47 P-INF (PSIA) 7.311 P01 (PSIA) 3.953 U-INF (FT/SEC) 3465 RHO-INF (SLUGS/FT3) 2.451E-03 MU-INF (LB-SEC/FT2) 7.923E-08 RE/FT (FI-1) 3.785E-06 MODEL STA (IN) 23.78 (X/L) 0.700

CH	POS	TAP	PR	PR/P01	Y(IN)	PML/PR	ML	REL (FI-1)	TL/T-INF	UL/U-INF	RHO-L/RHO-INF	MU-L/MU-INF	
1	3	1	9.093E-00	1.244E-00	.114	4.792E-01	1.081	9.249E-05	11.186	.452	4.48152	1.9957	8.088
2	3	2	1.115E-01	1.526E-00	.166	3.407E-01	1.251	1.155E-06	10.510	.507	4.6989	2.3824	7.763
3	3	3	1.255E-01	1.717E-00	.112	3.472E-01	1.351	1.305E-06	10.111	.537	4.8843	2.6224	7.566
4	3	4	1.354E-01	1.848E-00	.163	3.207E-01	1.419	1.416E-06	9.838	.556	5.0199	2.7930	7.428
5	3	5	1.478E-01	2.022E-00	.216	2.948E-01	1.495	1.547E-06	9.536	.577	5.1787	2.9892	7.274
6	3	6	1.525E-01	2.046E-00	.254	2.957E-01	1.522	1.597E-06	9.429	.584	5.2378	3.0612	7.218
7	3	7	1.623E-01	2.219E-00	.313	2.685E-01	1.579	1.703E-06	9.208	.599	5.3635	3.2129	7.103
8	3	8	1.720E-01	2.353E-00	.365	2.533E-01	1.636	1.815E-06	8.989	.613	5.4939	3.3684	6.987
9	3	9	1.784E-01	2.441E-00	.415	2.442E-01	1.669	1.883E-06	8.863	.621	5.5724	3.4612	6.919
10	3	10	1.867E-01	2.541E-00	.499	2.310E-01	1.724	2.000E-06	8.656	.634	5.7052	3.6170	6.808
11	3	11	2.032E-01	2.779E-00	.606	2.145E-01	1.794	2.168E-06	8.392	.651	5.8923	3.8341	6.657
12	3	12	2.165E-01	2.966E-00	.702	2.010E-01	1.866	2.333E-06	8.134	.665	6.0715	4.0400	6.519
13	3	13	2.318E-01	2.991E-00	.802	1.933E-01	1.874	2.352E-06	8.106	.667	6.0925	4.0638	6.504
14	3	14	2.435E-01	2.420E-00	.92	2.041E-01	1.849	2.290E-06	8.197	.662	6.0248	3.9865	6.555
15	3	15	2.025E-01	2.770E-00	.981	2.152E-01	1.796	2.164E-06	8.389	.650	5.8872	3.8283	6.561

CH	TC	TTR	TTR/TC	Y(IN)	PML/P01
1	1	1343	1.0015	.051	5.960E-01
2	2	1372	1.0231	.131	
3	3	1343	1.0164	.202	
4	4	1376	1.0261	.402	
5	5	1362	1.0157	.599	



GROUP 334 ALPHA-MODEL 50.03 SIA IXVL 0.700



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FIELD OF VIEW

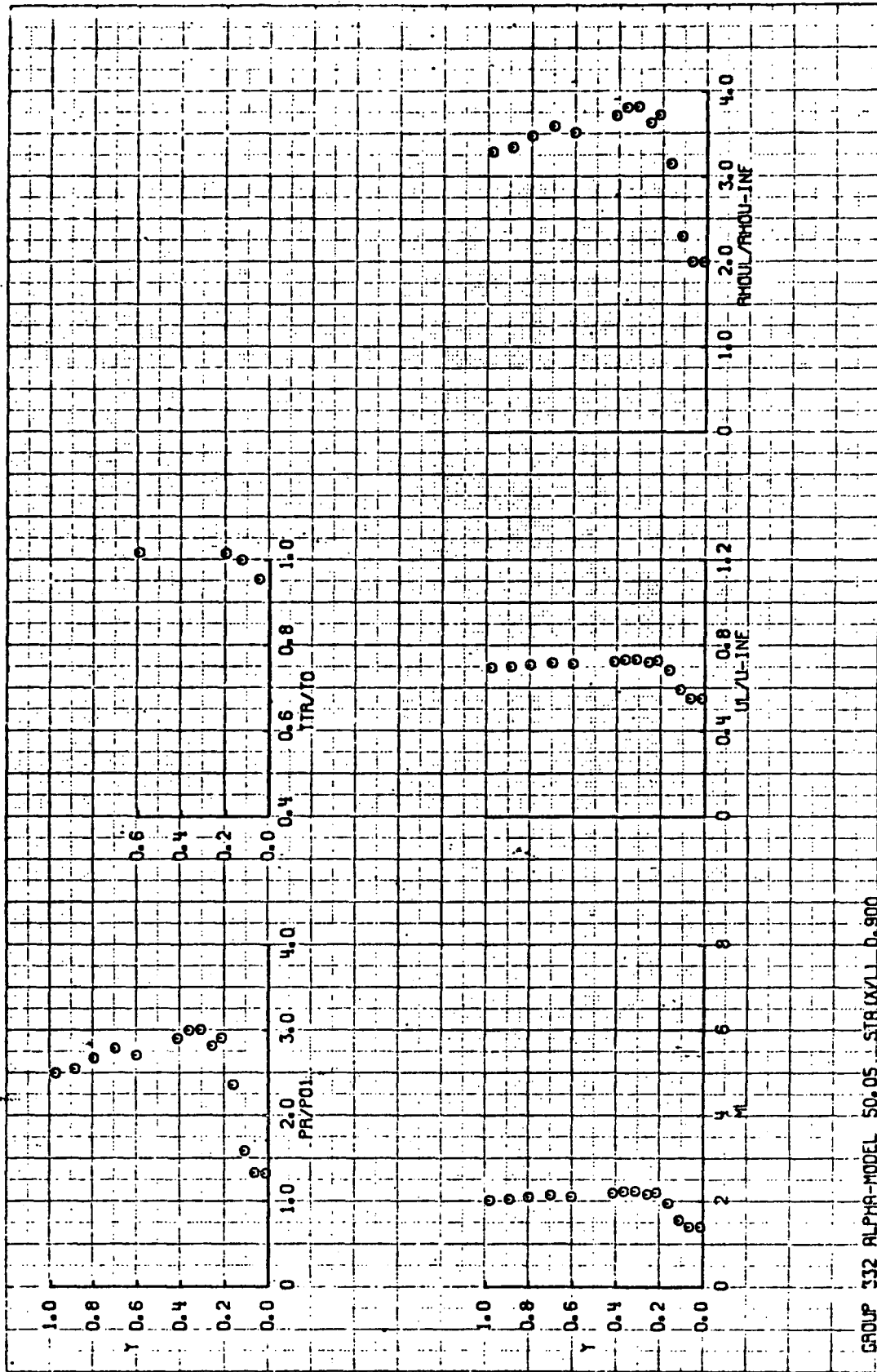
Group 334,  $\alpha = 50$ , RAKE STA (X/L) 0.7

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 AEDC (AO, INC.) ARNOLD AFS, TENNESSEE  
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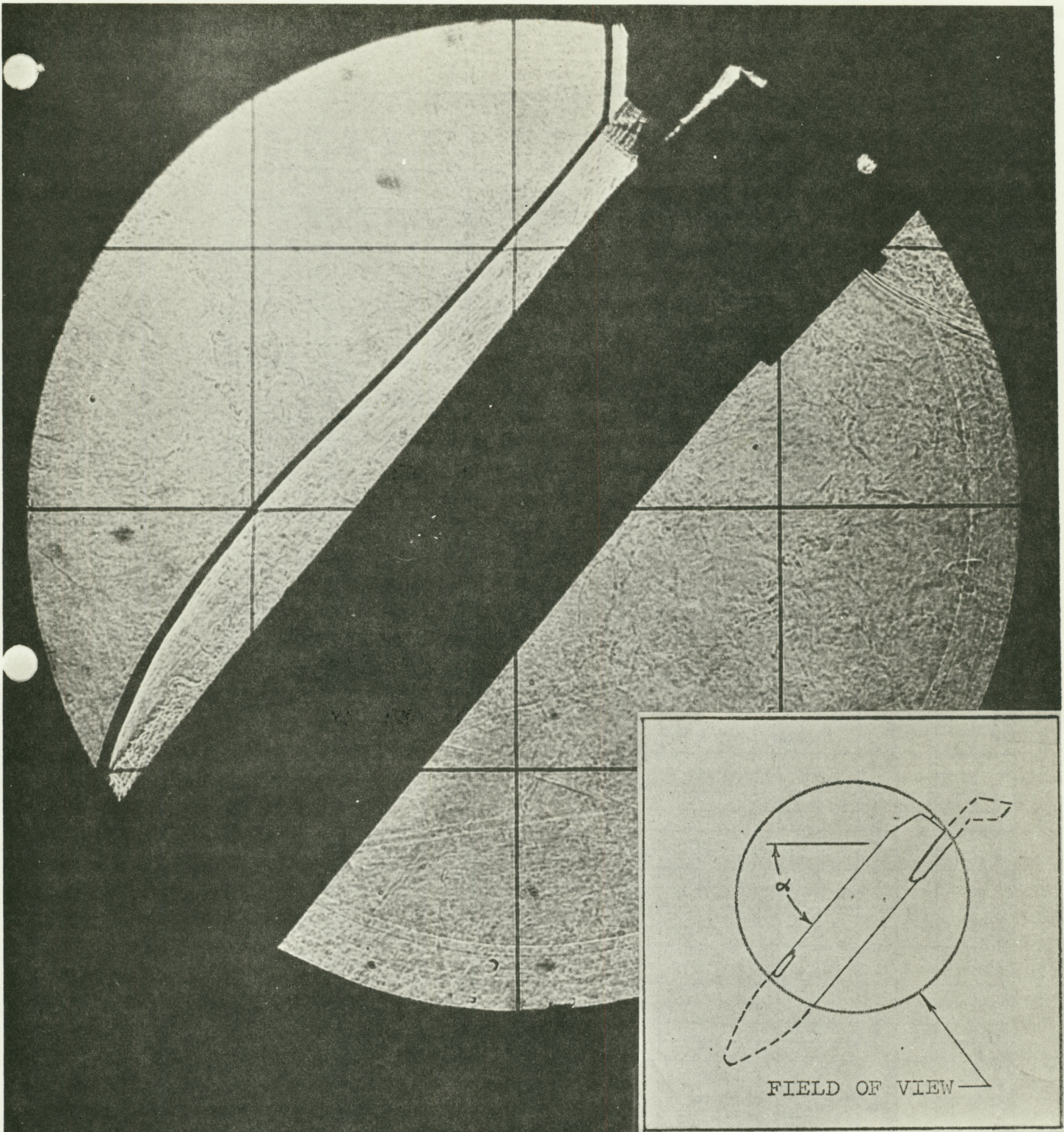
GROUP	CONFIG	MODEL	MACH NO.	PO PSIA	TO DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW		
332	42	M/DAC-H	8.00	860.8	1341	50.05	-0.06	-50.00	180.00	0.0		
Y-1NF (DEG R)	P-1NF (PSIA)	Q-1NF (PSIA)	U-1NF (FT/SEC)	RHO-1NF (SLUGS/FT3)	MU-1NF ((LB-SEC/FT2)	RE/FT (FT-1)	MODEL STA (X/L)	L (IN)				
97	8.82E-02	7.307	3.950	2.449E-03	7.823E-08	3.763E-06	.900	23.78				
CH POS	TAP	PH (PSIA)	PR/PO1	Y1(N)	PML/PH	ML (FT-1)	REF (FT-1)	TL/T-1NF	UL/U-1NF	RHO1/RHO-1NF	RHOUL/RHOU-1NF	MUL/MU-1NF
1	1	9.678E-00	1.325E-00	.014	3.269E-01	1.403	1.009E-06	9.901	.552	3.6250	2.0005	7.460
2	3	9.708E-00	1.329E-00	.066	3.259E-01	1.405	1.011E-06	9.893	.553	3.6269	2.0041	7.456
3	3	1.128E-01	1.594E-00	.112	2.733E-01	1.563	1.215E-06	9.269	.595	3.8711	2.3035	7.135
4	3	1.725E-01	2.361E-00	.163	1.894E-01	1.964	1.878E-06	7.791	.685	4.6055	3.1560	6.324
5	3	2.126E-01	2.910E-00	.216	1.488E-01	2.200	2.391E-06	7.012	.728	5.1173	3.7271	5.864
6	3	2.061E-01	2.821E-00	.258	1.575E-01	2.163	2.304E-06	7.129	.722	5.0331	3.6339	5.935
7	3	2.193E-01	3.005E-00	.313	1.441E-01	2.239	2.497E-06	6.890	.735	5.2074	3.8265	5.790
8	3	2.193E-01	2.397E-00	.265	1.445E-01	2.235	2.477E-06	6.902	.734	5.1983	3.8165	5.798
9	3	2.122E-01	2.904E-00	.315	1.491E-01	2.199	2.387E-06	7.018	.728	5.1128	3.7221	5.868
11	3	1.977E-01	2.704E-00	.206	1.800E-01	2.116	2.197E-06	7.290	.714	4.9288	3.5182	6.026
12	3	2.037E-01	2.788E-00	.202	1.553E-01	2.149	2.272E-06	7.173	.720	5.0024	3.6000	5.961
13	3	1.953E-01	2.673E-00	.202	1.670E-01	2.103	2.107E-06	7.324	.711	4.8988	3.4848	6.052
14	3	1.861E-01	2.544E-00	.192	1.700E-01	2.096	2.094E-06	7.512	.701	4.7766	3.3483	6.162
15	3	1.826E-01	2.499E-00	.181	1.733E-01	2.024	2.000E-06	7.594	.697	4.7311	3.2974	6.205

CH	TC	Y1M (DEG R)	Y1M/ZC (DEG R)	Y1(N)	PML/PO1
1	1	1279	9538	.051	9.330E-01
2	2	1340	9993	.131	
3	3	1361	10145	.202	
6	6	1364	10172	.599	

GROUP 332



GROUP 332 ALPHA-MODEL 50.05 STR (XV) 0.900



Group 332,  $\alpha = 50$ , RAKE STA (X/L) 0.9

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