

Cryogenic Thermal Conductivity Measurements on Candidate Materials for Space Missions

Jim Tuttle, Ed Canavan, and Amir Jahromi

NASA Goddard Space Flight Center, Code 552 Greenbelt, Maryland, 20771, USA



Introduction



- Many NASA missions include cryogenic instruments
- Spacecraft and instruments include optimized materials/assemblies
 - Highly-conductive annealed pure metals
 - Engineered materials
 - Polymers Alloys
 - Alloys
 - Composites
 - Ceramics
 - Customized electrical cables/harnesses
- Candidate materials often selected based on room temp. properties
- Often longitudinal cryogenic thermal conductivity is unknown
- We developed a thermal conductivity facility for JWST in 2004
- We have characterized ~ 30 samples since then





For one-dimensional heat flow in a material,

$$\frac{\dot{Q}}{A} = \kappa \frac{dT}{dx}$$

- κ : thermal conductivity [W/m/K]
- *q*: power [W]
- *A*: cross sectional area [m²]
- *T*: temperature [K]
- *x*: axial distance [m]
- Basic approach
 - Flow heat through sample
 - Measure temperature gradient
- We chose to perform "absolute" measurement
 - Relative measurements: lower precision



Simple Approach

Cryogenics and Fluids Branch

- Control base temperature
- Apply heat to sample's free end
- Measure (small) ΔT

 $\kappa(\bar{T}) = \frac{L\,\dot{Q}}{A\,\Delta T}$

- *L*: sample length [m]
- *A*: cross section [m²]
- $\overline{T} = (T_{\text{Sample}} + T_{\text{Base}})/2$





Complications



- Ohmic heating in heater leads
- Heat conducted in leads
- Heat radiated to surroundings

 $T^4_{Hot} - T^4_{Cold} \sim 4 \bar{T}^3 \Delta T$

- Joint resistance at base
- Joint resistance at floating end
- Absolute thermometer errors

$$\frac{\dot{Q}_{H} - \dot{Q}_{TL} - \dot{Q}_{HL} - \dot{Q}_{R}}{A} = \kappa(\bar{T}) \frac{(T_{F} - \delta T_{F}) - (T_{B} - \delta T_{B}) + \Delta T_{F} + \Delta T_{E}}{L}$$





Our Test Configuration

- Based on approach described in 1973 Moore, Williams and Graves RSI paper
- Guard surrounds sample: Controlling $T_{\text{Guard Top}} = T_{\text{Sample Top}}$ reduces sample heat radiation
- "Fiberfrax" insulation eliminates remaining sample radiation
- Intermediate thermometers eliminate joint resistance effect
- Optimizing sample heater and leads minimizes ohmic heating in leads
- Lead heat-sinking minimizes lead heat conduction





Instrumentation



- Thermometers
 - LakeShore Cryotronics SD-package CernoxTM sensors
 - Calibrated (resistance vs. *T*) from 1 to 325 K
- Heaters
 - Sample heater is $10 \text{ K}\Omega$ metal-film resistor
 - Leads: size, material chosen to give round-trip resistance less than $\sim 10 \Omega$ inside guard
 - Base and guard heaters: 50Ω
 - made by winding stainless steel wire around flange
 - we don't measure the power for these heaters
- Temperature readout/control boxes
 - Cryogenic Control Systems Cryocon Model 32B Controller
- Heater voltage and current readout
 - Keithley Model 2000 6.5-digit multi-meters



Data Acquisition and Analysis

- For each value of $\overline{T} = (T_{\text{Sample}} + T_{\text{Base}})/2$:
 - Perform 4 different steady-state "balances"
 - For each balance, control $T_{\text{guard}} = T_{\text{Sample}} > T_{\text{Base}}$
 - Measure $\Delta T = T_{\text{Far}} T_{\text{Near}}$
 - Measure \dot{Q} = sample control power

$$\kappa(\bar{T}) = \frac{L}{A} \frac{d\dot{Q}}{d\Delta T}$$

- To first order, differential measurement eliminates effect of absolute temperature errors
 - $\frac{d\dot{Q}}{d\Delta T}$ is more accurate than any single $\frac{\dot{Q}}{\Delta T}$ value
 - Least-squares fit of 4 different ΔT values provides statistical uncertainty in $\frac{d\dot{Q}}{d\Delta T}$





- Thermometer R vs. T calibrations have "scatter" due to measurement uncertainty
- Assume that "true" R(T) is a smooth function approximated by a smoothing fit
 - LakeShore Cryotronics provides smoothing Chebyshev Polynomial fits
 - We performed cubic spline smoothing fit on a cal. curve
- Our readout box uses cubic spline interpolation to get *T* from *R*
 - Interpolation forces curve to go through every "scattered" point
 - Causes local dR/dT errors relative to slope of "true" smooth curve
 - A local error in dR/dT results in a proportional local error in κ





- Graphed slope difference between spline-smoothed curve and spline interpolations:
- Blue curve: interpolation of raw calibration points
- Red curve: interpolation of Chebychev fit points
- Above 6 K, raw points give max. slope error of 0.3% (mostly below 0.2%)
- Improvement is possible by loading Chebychev fit points into readout box





- To first order, keeping $T_{\text{Sample}} = T_{\text{Guard}}$ eliminates effect of sample-guard heat leaks
 - For small ΔT values, T_{Sample} T_{Guard} calibration curve mismatches are assumed constant for balances with a given \overline{T}
 - Constant mismatches result in constant sample-guard heat leak
 - This does not effect $\frac{d\dot{Q}}{d\Delta T}$
- However, Fiberfrax effective thermal conductivity has a strong (T^3) temperature dependence
- We performed finite-element thermal model to evaluate second order effects in $\frac{d\dot{Q}}{d\Delta T}$



- Worst-case error at 300 K
- PVC has very low $\kappa = 0.16$ W/m/K at 300 K
- Modeled error vs. sample diameter inside 32 mm guard
- It's best to make sample diameter as large as practical
- This error is proportional to $1/\kappa$, so much lower for other materials





High Conductivity Samples





Medium Conductivity Samples



Cryogenics

and Fluids Branch





S-glass [(45,-45,0)4s layup, S2 glass in EX1522 epoxy matrix]
T300 [(45,0,-45)2s layup, T300 carbon fiber with 5HS weave in RS-3C epoxy matrix]







- It's not too difficult to perform high-precision thermal conductivity measurements between 4 K and room temperature
- NASA/GSFC's cryogenics group is equipped to perform such measurements for customers at any NASA center
- Thanks to the James Webb Space Telescope program, which funded the development of the technique and facility