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# Cooling the Origins Space Telescope

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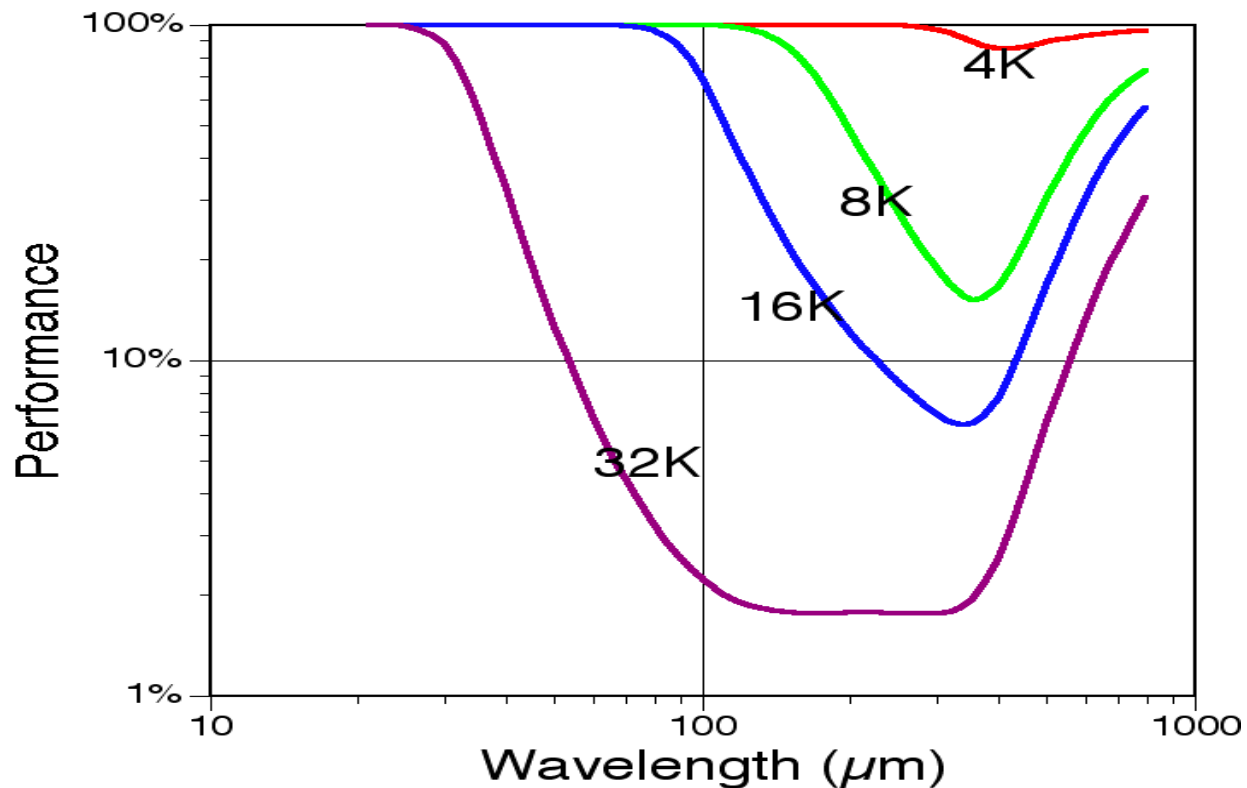
# What is OST?

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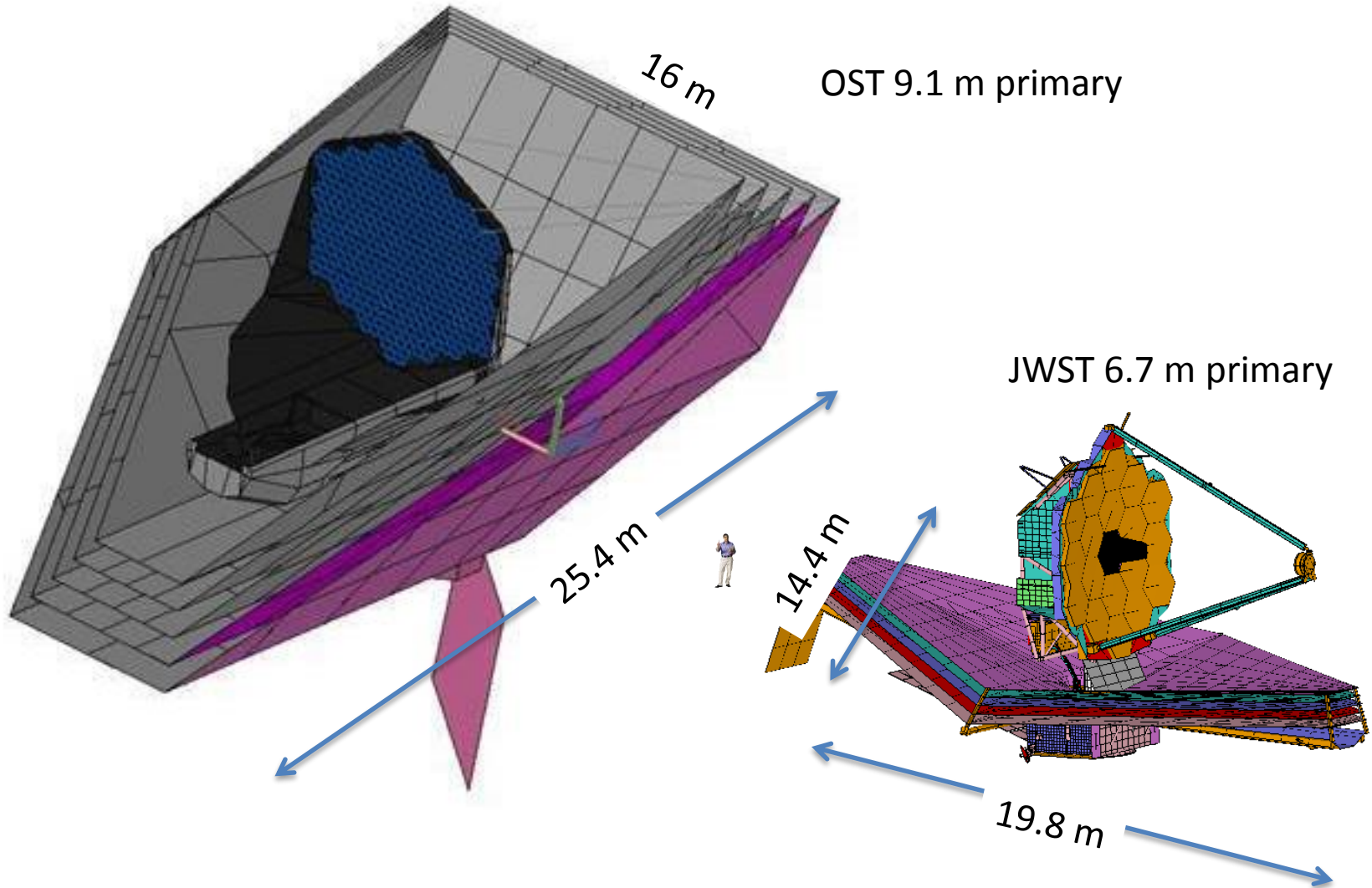
- NASA Headquarters Astrophysics Division commissioned 4 studies for one possible flagship mission to launch in the 2030's
  - Lynx (X-ray Surveyor)
  - LUVOIR (Large UV, Optical, and near InfraRed Observatory)
  - HabEx (Habitable Exoplanet mission)
  - Origins Space Telescope (OST) (Far IR Surveyor)

# Why Do We need a 4.0 K Telescope?

- OST will cover the wavelength range from 6  $\mu\text{m}$  to 600  $\mu\text{m}$ 
  - The goal is to be background limited – limited by the cosmos rather than self emission from the telescope



# How Big is This Thing?



# One Question

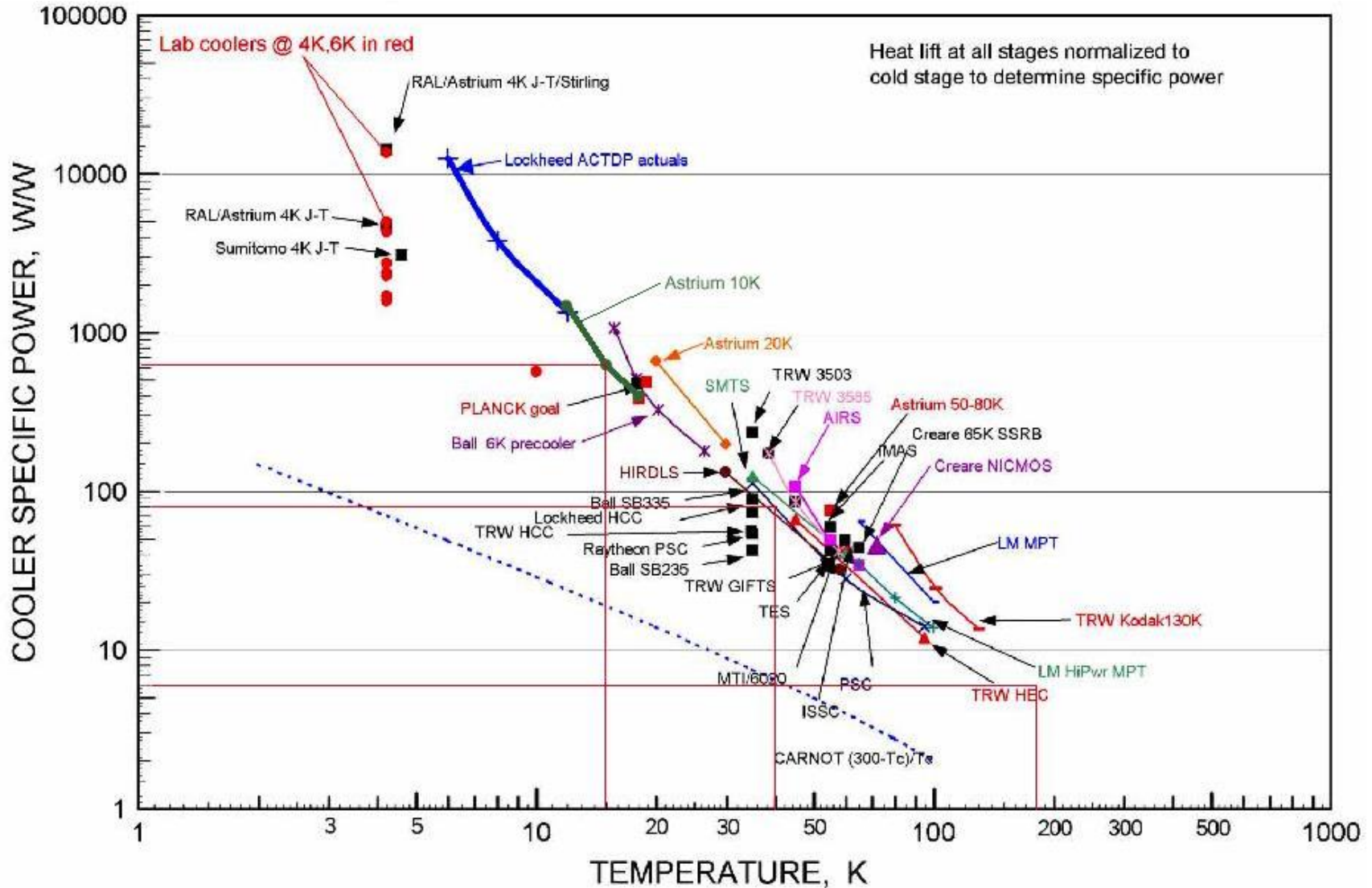
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- How can we do a 9 m diameter 4 K telescope when a 6.7 m 40 K telescope (JWST) was so difficult?
  - Cryocoolers are now more mature
    - Cryocoolers at low temperature have a huge advantage over radiative cooling at low T
  - OST has a longer wavelength so the optics are less challenging
  - Low temperature has advantages
    - Low thermal contraction with changing temperatures for one
  - The design is driven by cryo/thermal considerations
  - Cryogenics leads to solutions!!!

# State of the Art for cooling

- ACTDP and follow up matured coolers from 3 different companies
- Approximately 10,000 W of input power per W of cooling power
- Space Cryocooler Reliability is Extremely High
  - From Ron Ross's ongoing survey

# Cooling Power Vs. T



# Staged Cooling

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- Utilize staged cooling to go from 300 to 0.05 K
  - Start with radiative cooling of the sunshield
  - Use 3 stages of cryocooler cooling
  - Finish with a subKelvin cooler to provide 50 mK



# Nominal Heat Flows

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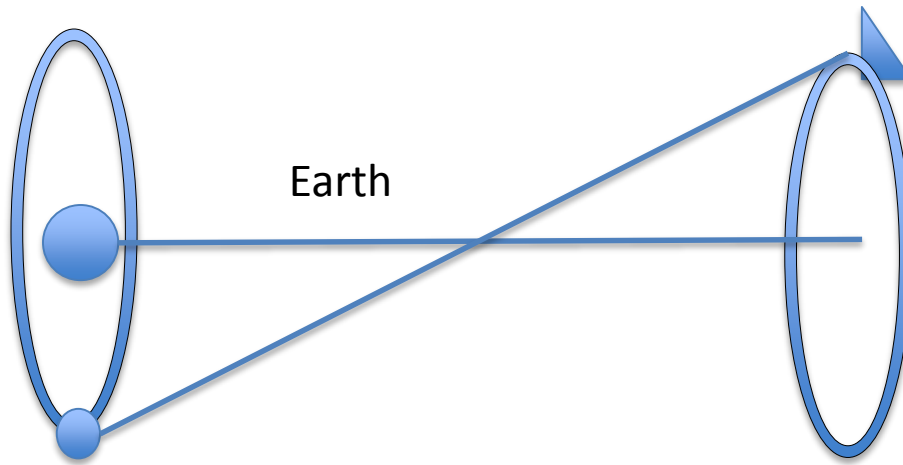
- Rough calculation of the heat absorbed by the 4 K cryocooler stage is broken down as follows:
  - Telescope: radiation: 55 mW, conduction from structure 20 mW, conduction from harnesses 30 mW
  - Instrument dissipation: Maximum 100 mW mainly from low temperature preamplifiers
- Will use eight 50 mW cryocoolers in parallel which provides redundancy, 100% margin on the expected cooling load, and keeps size close to current technology

# Sunshield Principle

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- Radiate horizontally, block radiation perpendicular
- Practical Considerations
  - Solar pressure imbalance
  - Deployment
  - The Sun, Earth, and Moon do not stay in one place

# Keep Out Angles



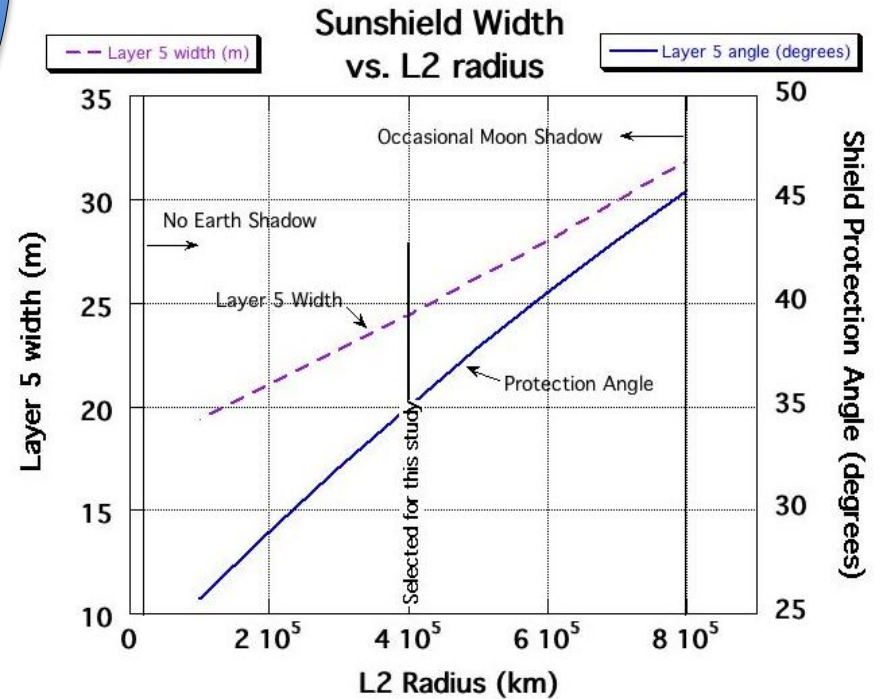
OST

L2 Orbit

Earth

Moon

Field of Regard adds to shield size  
 OST Pitch =  $+5^\circ/-45^\circ$ , Roll =  $\pm 5^\circ$ ,  
 Yaw =  $360^\circ$

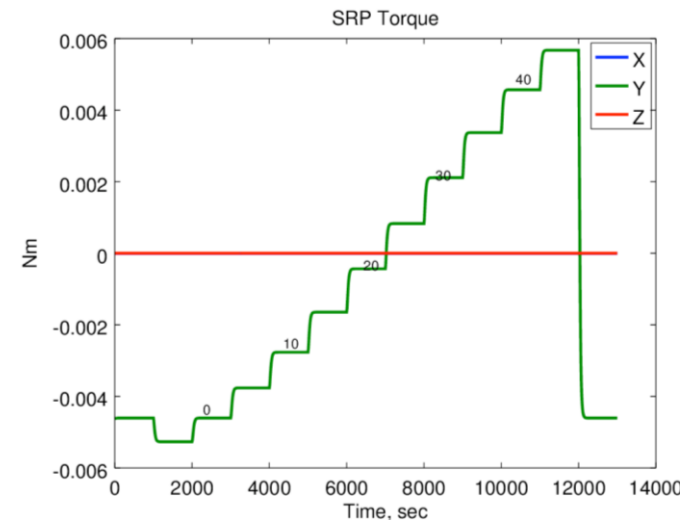
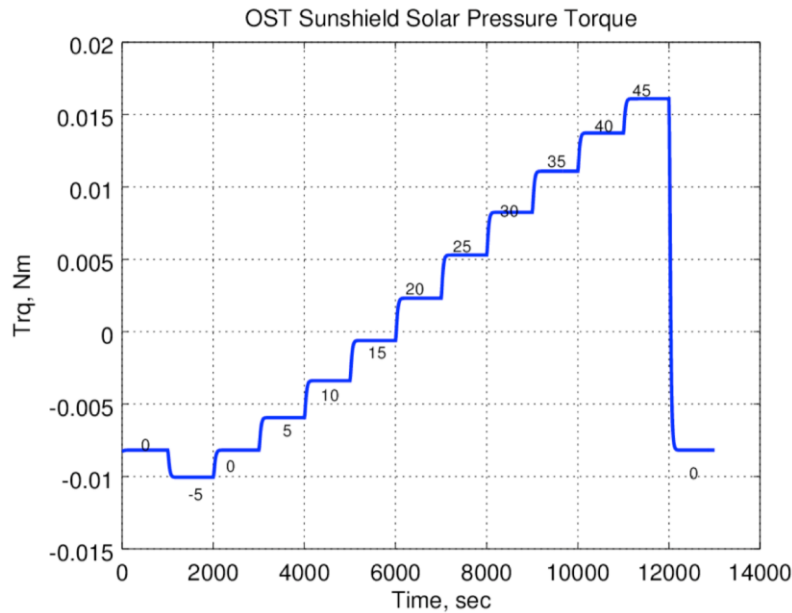
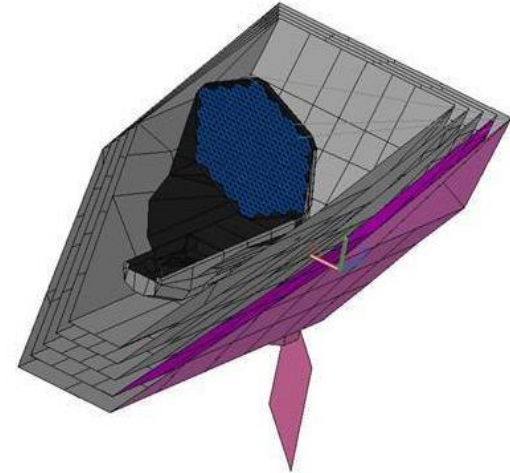
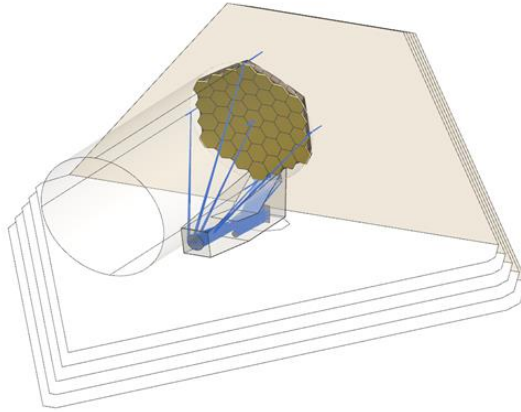


# Solar Pressure

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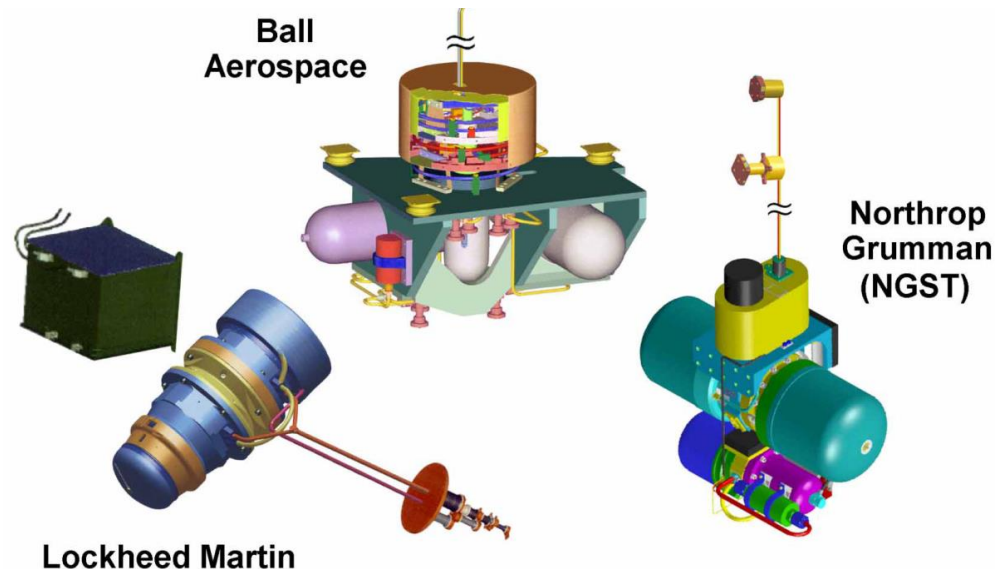
- Ideally the center of solar pressure ( $\sim 9 \mu\text{Pa}$ ) and center of mass are in the same place
- Any offset must be overcome with momentum wheels and propulsion
  - Aside from mass, frequent propulsive maneuvers disrupt observing time
- [Two Cases]

# Solar Torque - Two Cases



# Cryocooler Considerations

- Staged heat extraction
- Vibration issues
  - Not so much for image stabilization as for microphonics on detectors
- Packaging and distributing cooling
- 50 mW cooling at 4 K plus 20 K and 70 K cooling for 500 W of input power

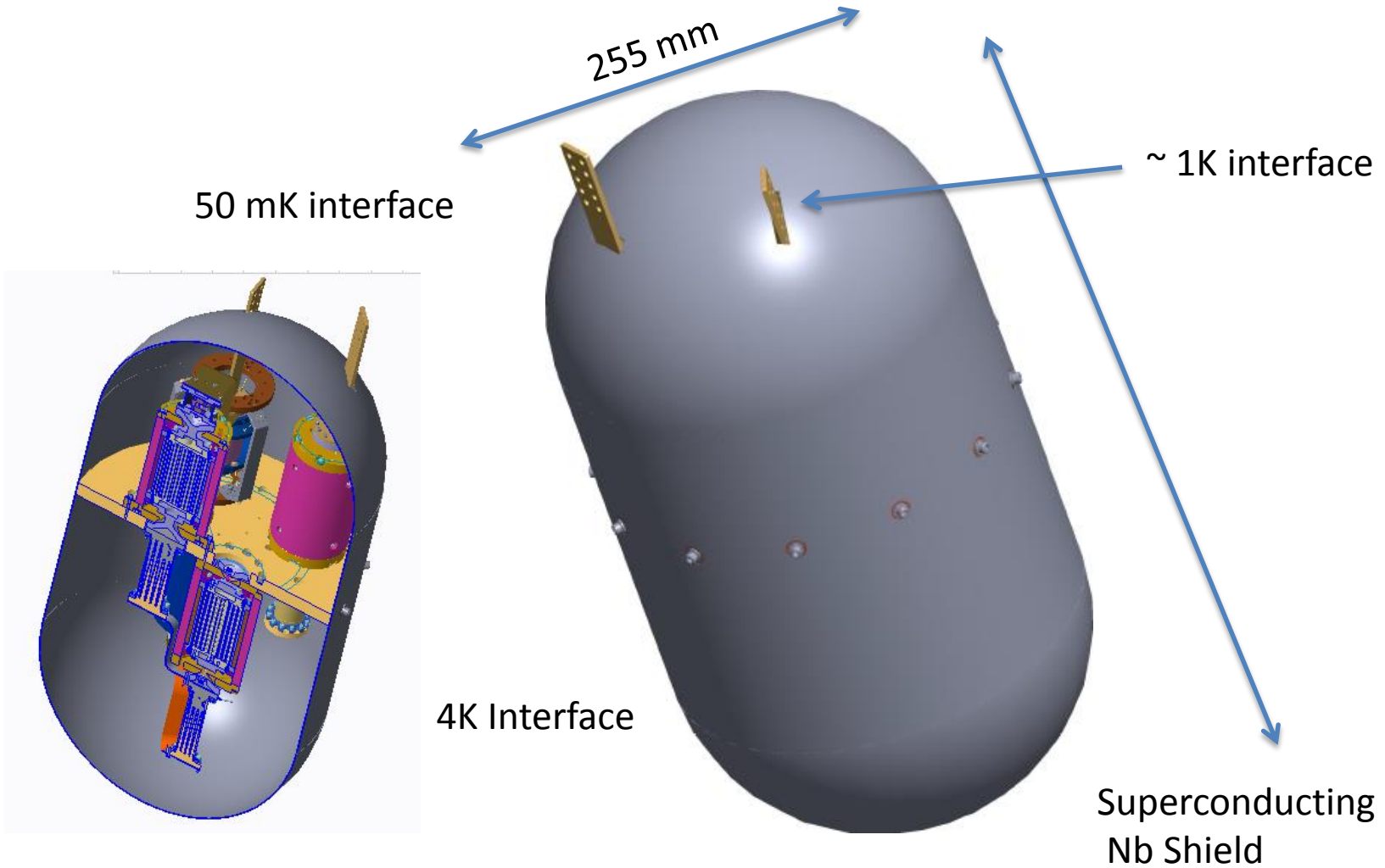


# Sub-Kelvin Cooling

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- Instruments whose detectors require cooling to less than 1 K will be sized to accommodate a provisional sub-Kelvin cooler. Such a cooler is currently at TRL4 and will be at TRL6 by the end of the current SAT (end of 2019)
  - Up to 5  $\mu$ W continuous cooling at 50 mK (max duty cycle)
  - Up to TBD mW cooling at  $\sim$ 1K
  - Heat rejection to 4 K
    - 6 mW (at max duty cycle), 3 mW at min duty cycle
  - Magnetic shielding to provide  $< 1\mu$ T at the boundary of the cooler
  - Mass  $\sim$  21 kg
  - Volume (see graphic)

# Multi-Stage ADR Sub-Kelvin Cooling





# Summary and Conclusions

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- The Origins Space Telescope is being studied as a flagship class astrophysics mission for the 2030's
  - 9.1 m diameter primary operating at 4 K
    - Cooling is achieved by proper staging of radiative coolers (sunshields), mechanical cryocoolers
  - 4 K and lower instruments
    - 3 instruments require subKelvin (50 mK) temperatures which could be provided by new continuous adiabatic demagnetization refrigerator