



Evaluation of Alternative Altitude Scaling Methods for Thermal Ice Protection System in NASA Icing Research Tunnel

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Outline

- Introduction
- Experimental Methods
- Results and Discussion
- Conclusion



Introduction

Motivation for Study

- Most thermal Ice Protection System (IPS) development and testing done in atmospheric icing tunnels that cannot simulate altitude effects.
- Altitude scaling is required to test IPS in atmospheric wind tunnels.
- Re-based scaling methods with empirical corrections typically used.
- A more robust scaling methods desired for development of current and future generation aircraft
- Joint NASA and NRCC research program conducted to study the issue.



Introduction

NRC AIWT Tests (2012, 2014, 2015)

- 18" chord NACA 0018 model with simple heated air IPS
- Re-based scaling method found to be inadequate
 - Airfoil surface temperatures well matched between altitude and ground conditions.
 - Accreted ice mass much larger than reference
 - Ice formed farther downstream than reference
 - Visual evidence of water re-entrainment into air flow.
- Two Weber number based method for scaling investigated.
 - One method matched We_{DW} (water density based)
 - Other method matched We_{DA} (air density based) and m_w/m_e (defined as Pi3).



Introduction

AIWT Tests (2012, 2014, 2015)

- We-based scaling methods produced ice accretions much closer in size and location to altitude reference conditions than Re-based scaling method.
- Provided means of altitude scaling based on primary physics and not empirical corrections.



Reference



Re Scaling



WePi3 Scaling



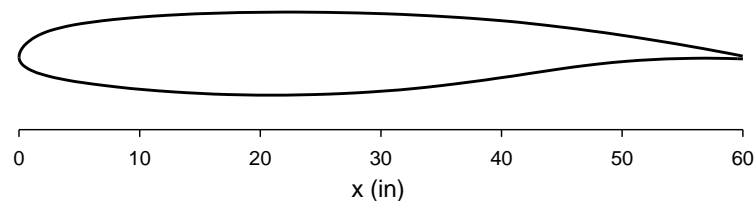
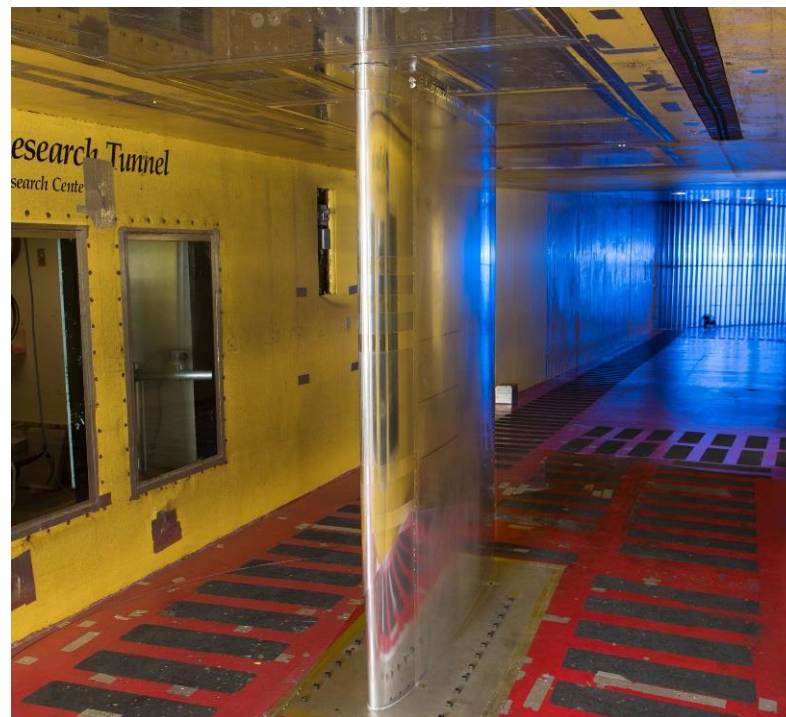
Introduction

2016 IRT Test

- Compared different scaling methods with a much larger business jet airfoil model.
- Ice protection system more similar to what is used on commercial aircraft.
- IRT cannot simulate altitude conditions.
- Results of different scaling methods can be compared to one another and trends compared to AIWT results.

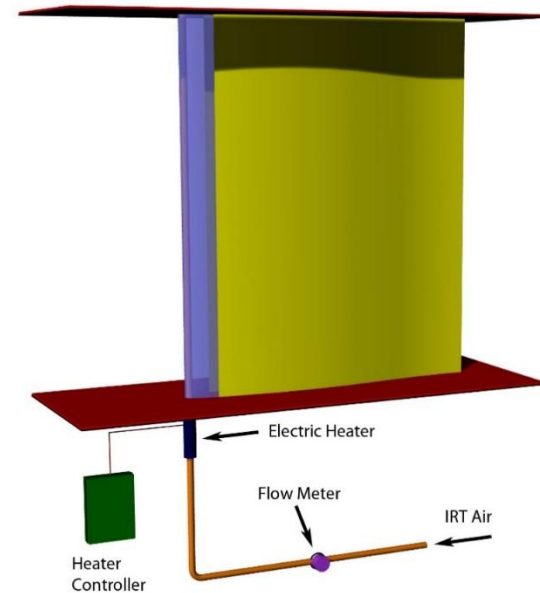
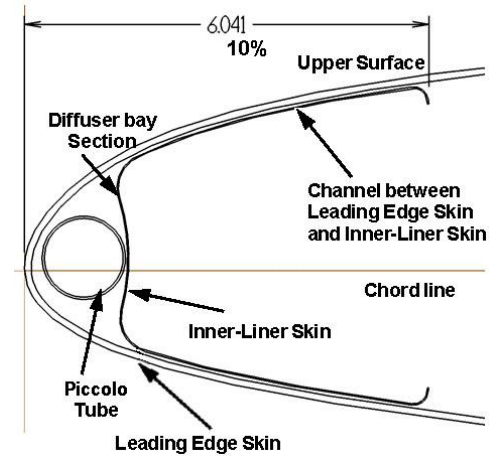
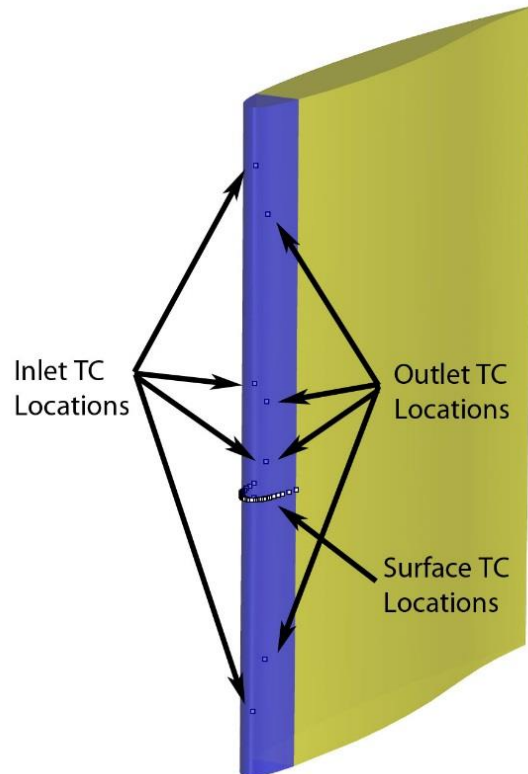
Experimental Procedure

- Test conducted in Icing Research Tunnel at NASA Glenn Research Center.
- 60 in. chord model representative of modern business jet.
- Piccolo tube IPS.
- Built for 2006 Wichita State University IPS analysis and modelling study.
- Extensively instrumented with temperature and pressure sensors.



Experimental Procedure

- Surface temperatures (32 TCs)
- 4 TCs in piccolo tubes (Inlet T)
- 4 TCs in diffuser liner (Outlet T)





Reference Conditions

| Flight Phase | Altitude (ft) | V (kts) | α (deg) | T_s (deg C) | T_t (deg C) | LWC (g/m ²) | MVD (μ m) |
|--------------|------------------|------------|-------------------|------------------|------------------|----------------------------|-------------------|
| Descent | 10000 | 180 | -1 | -14.2 | -10 | 0.35 | 19.1 |
| Cold Hold | 15000 | 180 | 3 | -20.1 | -15.8 | 0.24 | 17.5 |
| Warm Hold | 15000 | 180 | 3 | -8.6 | -4.3 | 0.49 | 17.4 |



Scaling Parameters

| Scaling Method | Scaling Parameters Held Constant | | | |
|----------------|----------------------------------|-------|-------|-------|
| Re | Re_{2r} | m_w | K_0 | T_r |
| WePi3 | WeDA | Pi3 | K_0 | T_r |
| WeDW | WeDW | m_w | K_0 | T_r |

- Required 2 step process to obtain scaled conditions
 - Run at Re-scaled conditions to obtain L.E. temperatures.
 - Run at We-scaled conditions with IPS adjusted to match the L.E. temperatures obtained at Re-scaled conditions.

Descent Scenario

| Scale Method | Alt (ft) | V (kts) | T_s (deg C) | LWC (g/m ³) | MVD (μ m) | m_w (g/m ² s) | Re_{2r} ($\times 10^6$) | WeDA | Pi3 | WeDW ($\times 10^6$) | Ice Mass (g) |
|-----------------|----------|---------|---------------|-------------------------|----------------|----------------------------|-----------------------------|------|------|------------------------|--------------|
| <i>Referenc</i> | 10000 | 180 | -14.2 | 0.35 | 19.1 | 17.6 | 0.224 | 5814 | 1.6 | 6.21 | N/A |
| Re | 1066 | 133 | -12.7 | 0.48 | 22.8 | 17.6 | 0.224 | 4315 | 2.24 | 3.38 | 20 |
| WePi3 | 1439 | 159 | -13.5 | 0.35 | 19.4 | 14.4 | 0.265 | 5814 | 1.62 | 4.84 | 8.5 |
| WeDW | 1782 | 180 | -14.2 | 0.34 | 21.6 | 17.6 | 0.297 | 7769 | 1.88 | 6.21 | 12.7 |



Re Scaling



WePi3 Scaling

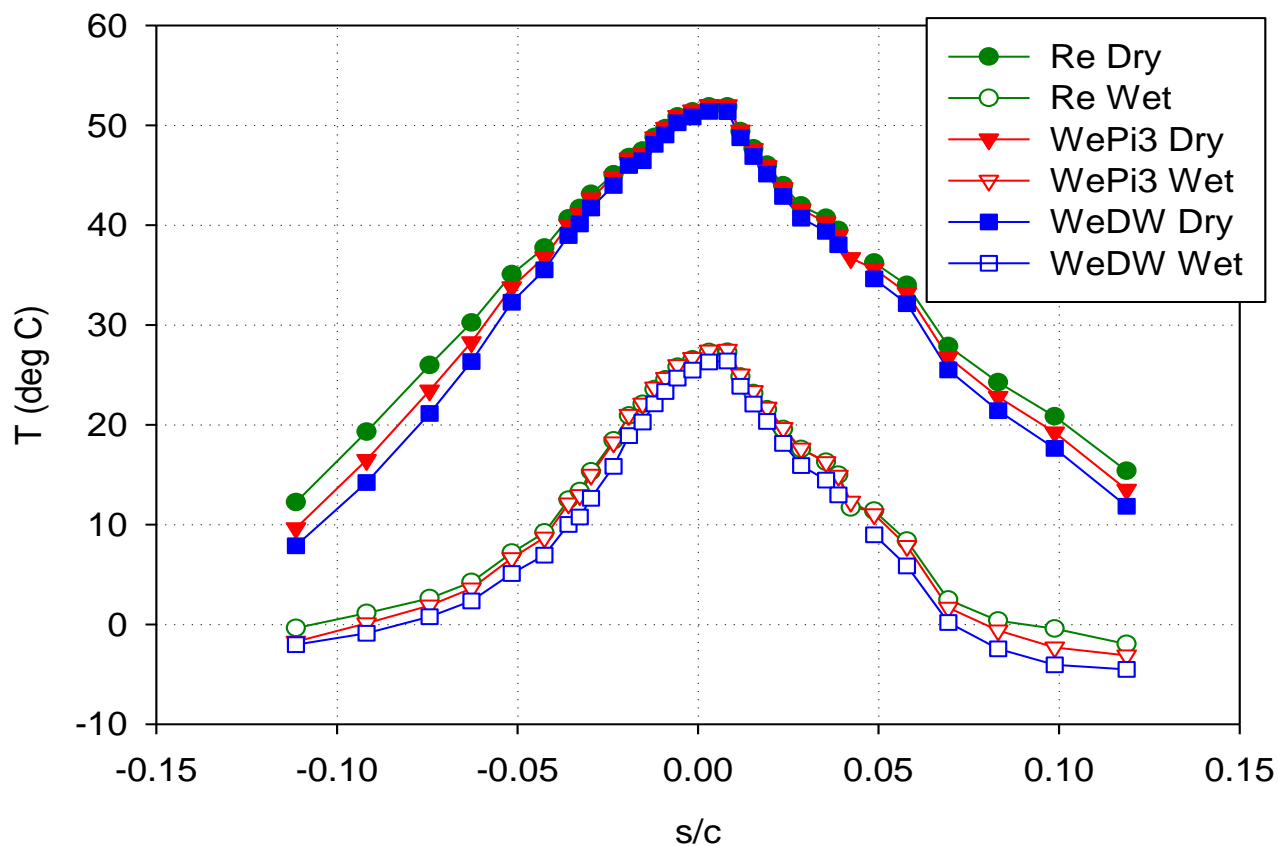


WeDW Scaling



Descent Scenario

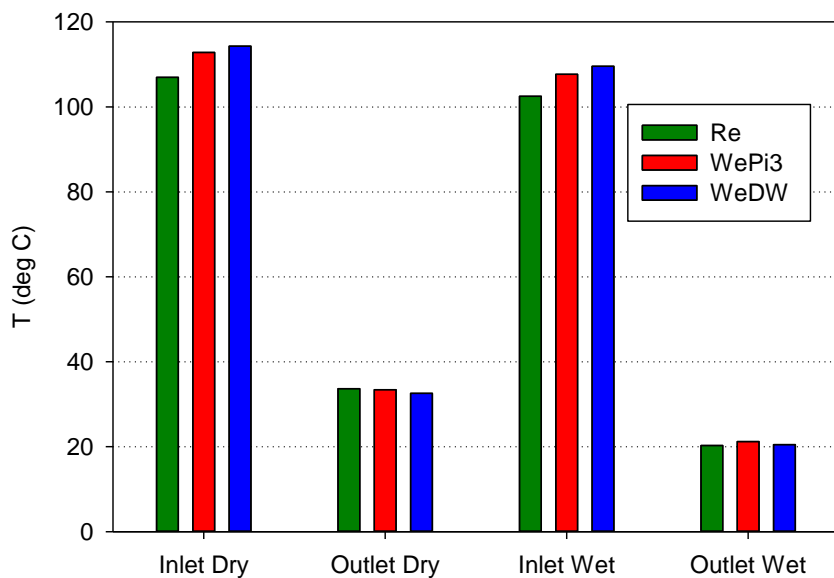
LE Surface Temperature



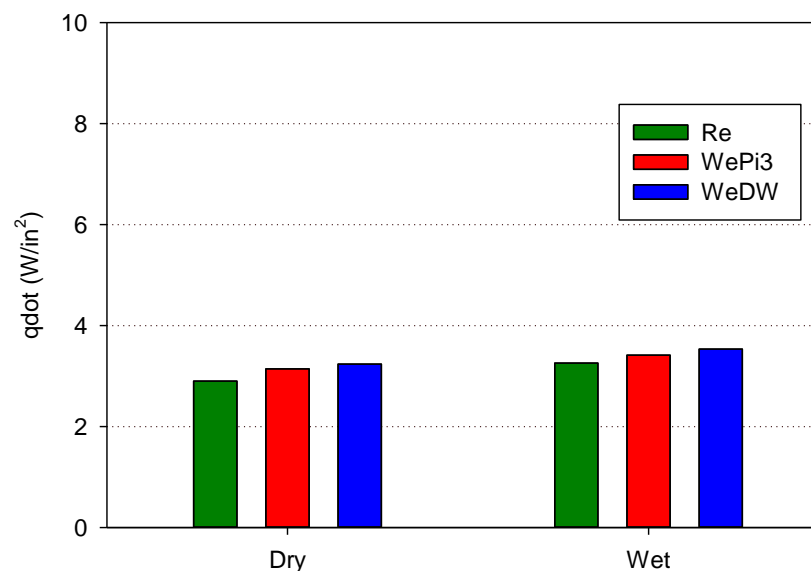


Descent Scenario

Heated Air Temp.

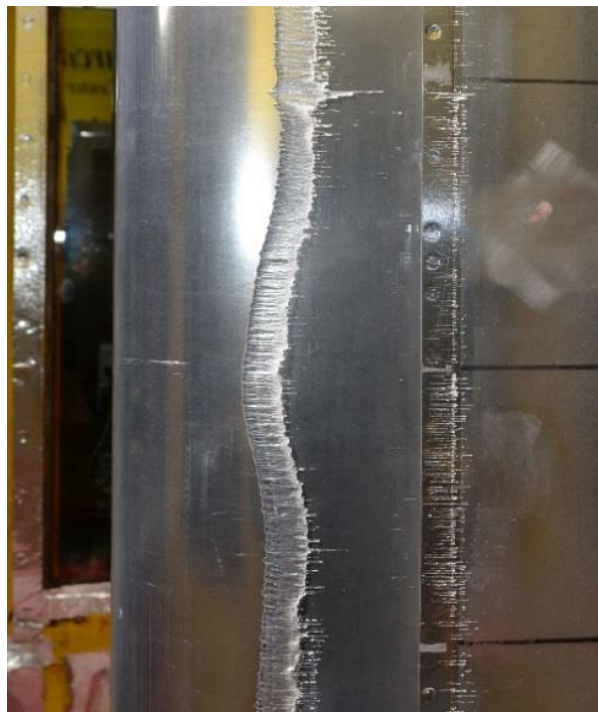


Heated Air Energy Input



Cold Hold Scenario

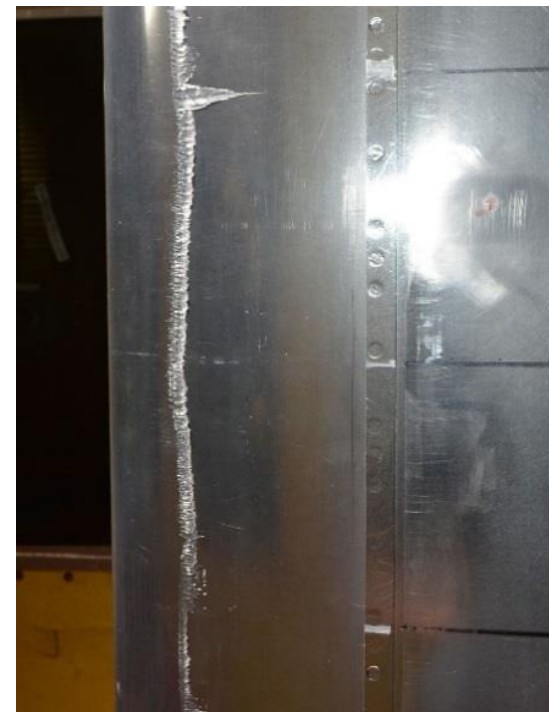
| Scale Method | Alt (ft) | V (kts) | T_s (deg C) | LWC (g/m^3) | MVD (μm) | m_w ($\text{g}/\text{m}^2\text{s}$) | Re_{2r} ($\times 10^6$) | WeDA | Pi3 | WeDW ($\times 10^6$) | Ice Mass (g) |
|------------------|----------|---------|---------------|-------------------------------|-----------------------|---|------------------------------------|------|------|------------------------|--------------|
| <i>Reference</i> | 15000 | 185 | -20.1 | 0.31 | 14.6 | 13.4 | 0.193 | 5147 | 1.06 | 6.54 | N/A |
| Re | 976 | 109 | -16.4 | 0.52 | 20.1 | 13.4 | 0.193 | 3065 | 1.87 | 2.27 | 54.2 |
| WePi3 | 1495 | 149 | -16 | 0.28 | 16.9 | 9.5 | 0.253 | 5147 | 1.06 | 4.22 | 5.0 |
| WeDW | 2087 | 185 | -15.7 | 0.27 | 18.9 | 13.3 | 0.31 | 8346 | 1.37 | 6.54 | 16.4* |



Re Scaling



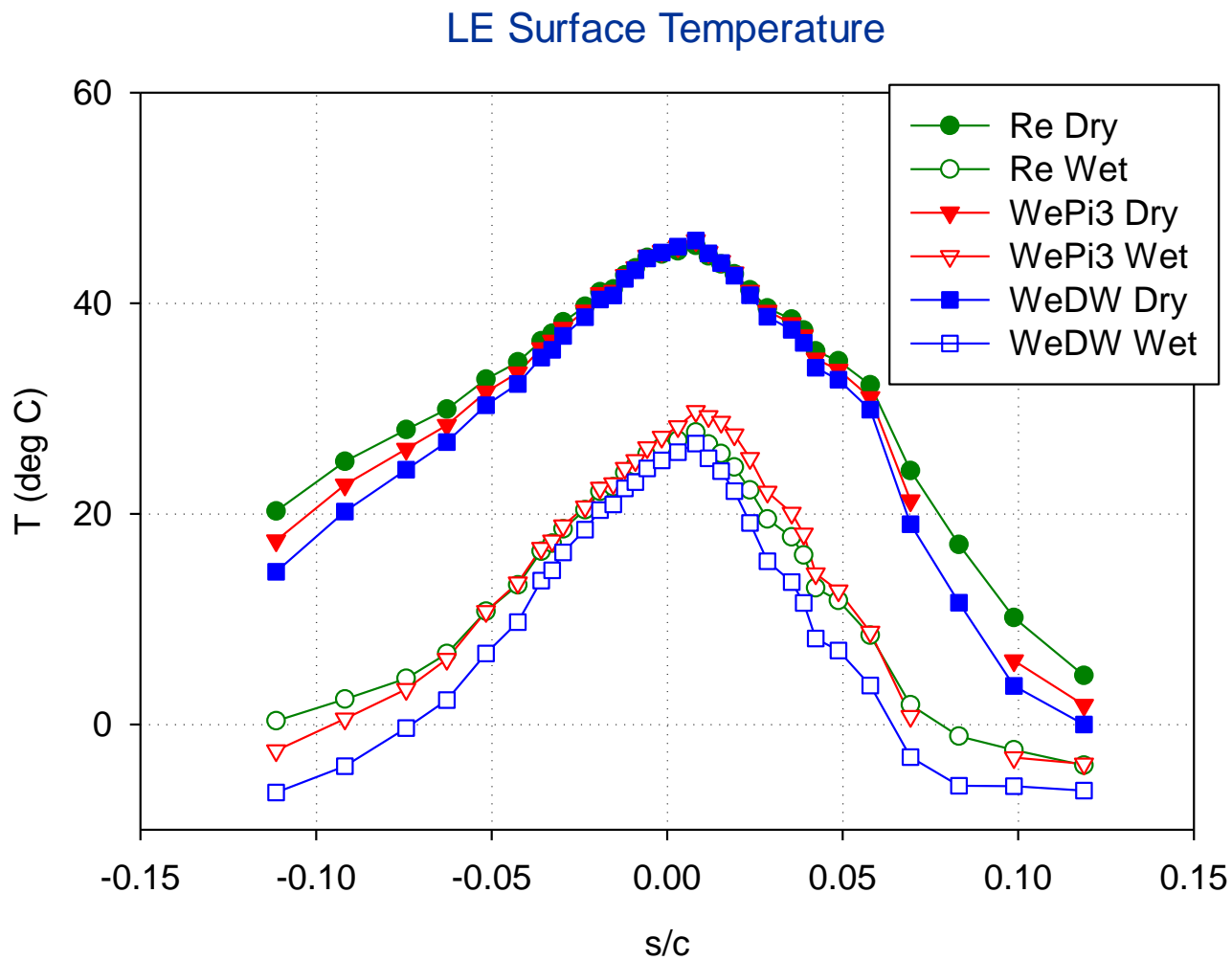
WePi3 Scaling



WeDW Scaling



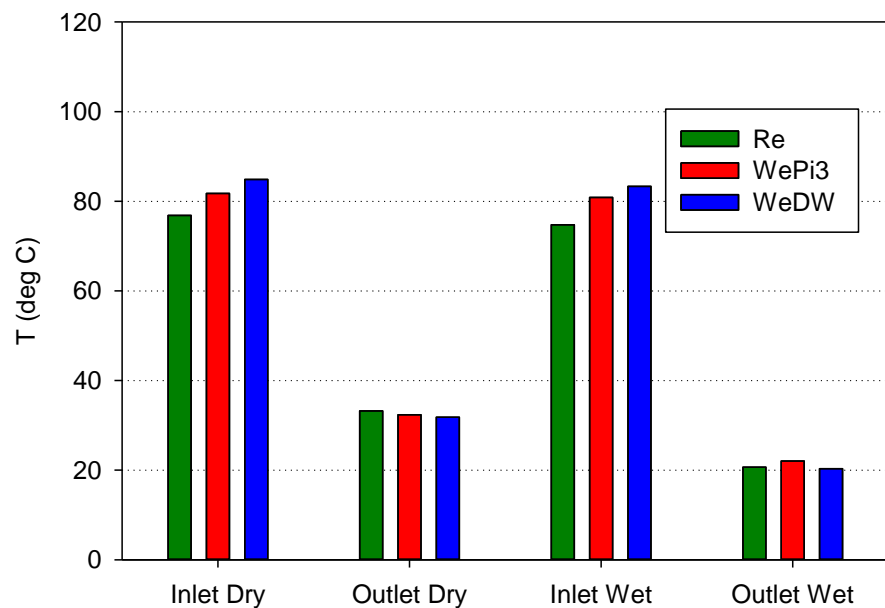
Cold Hold Scenario



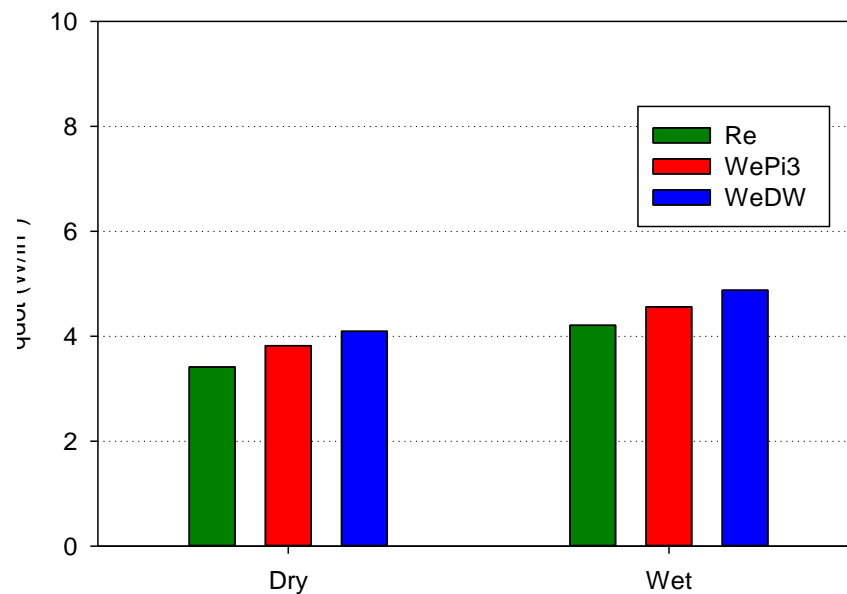


Cold Hold Scenario

Heated Air Temp.

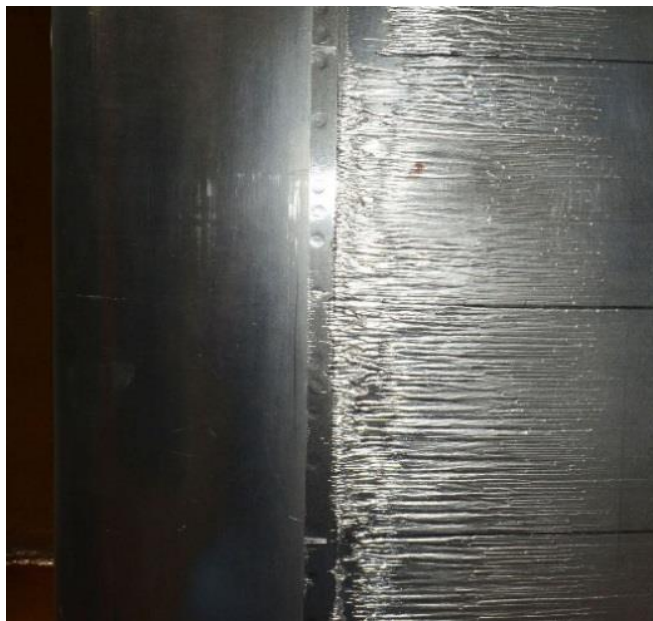


Heated Air Energy Input

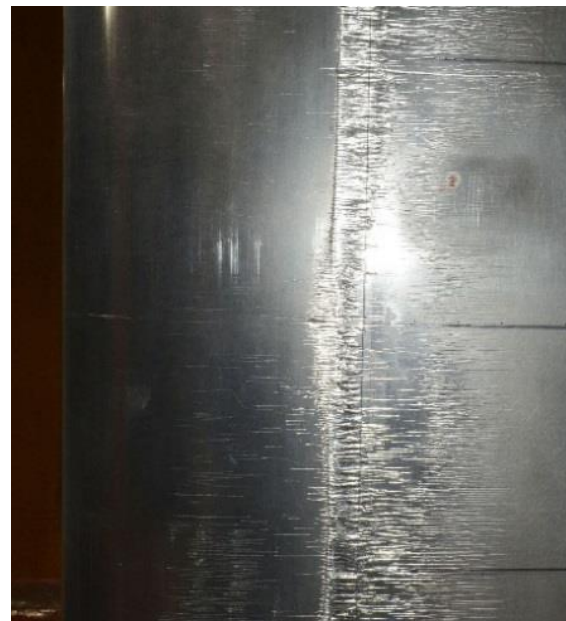


Warm Hold Scenario

| Scale Method | Alt (ft) | V (kts) | T_s (deg C) | LWC (g/m ³) | MVD (μm) | m_w (g/m ² s) | Re_{2r} (x10 ⁶) | WeDA | Pi3 | WeDW (x10 ⁶) | Ice Mass (g) |
|--------------|----------|---------|---------------|-------------------------|----------|----------------------------|-------------------------------|------|------|--------------------------|--------------|
| Reference | 15000 | 185 | -8.6 | 0.39 | 18.3 | 20.0 | 0.186 | 4922 | 1.74 | 6.55 | N/A |
| Re | 1336 | 109 | -6.3 | 0.66 | 25.4 | 20.0 | 0.186 | 2923 | 3.09 | 2.26 | 207.5 |
| WePi3 | 1814 | 147 | -7.3 | 0.36 | 19.6 | 13.6 | 0.241 | 4922 | 1.74 | 4.12 | 64.5 |
| WeDW | 2454 | 184 | -8.6 | 0.37 | 22 | 20.0 | 0.299 | 8005 | 2.24 | 6.55 | 138.8 |



Re Scaling



WePi3 Scaling

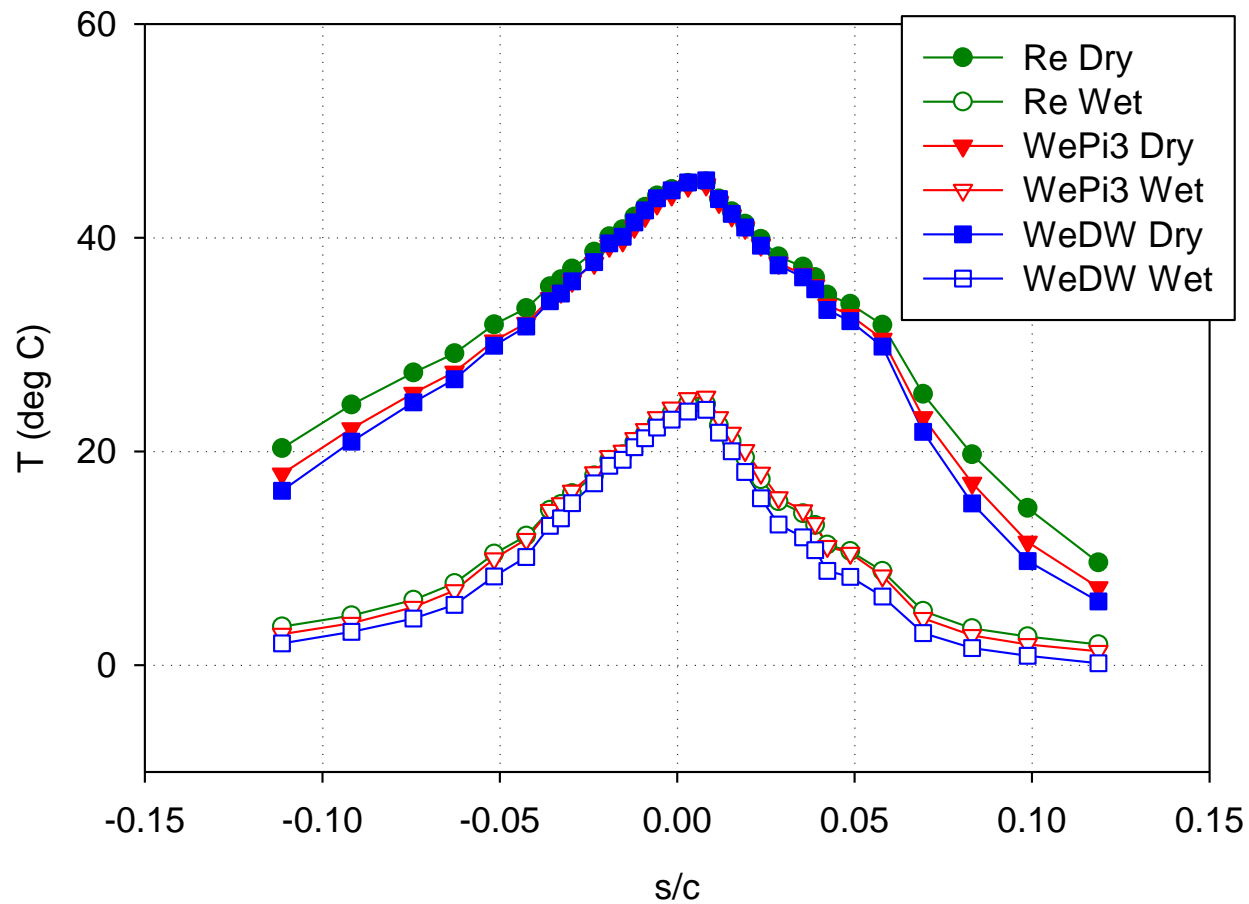


WeDW Scaling



Warm Hold Scenario

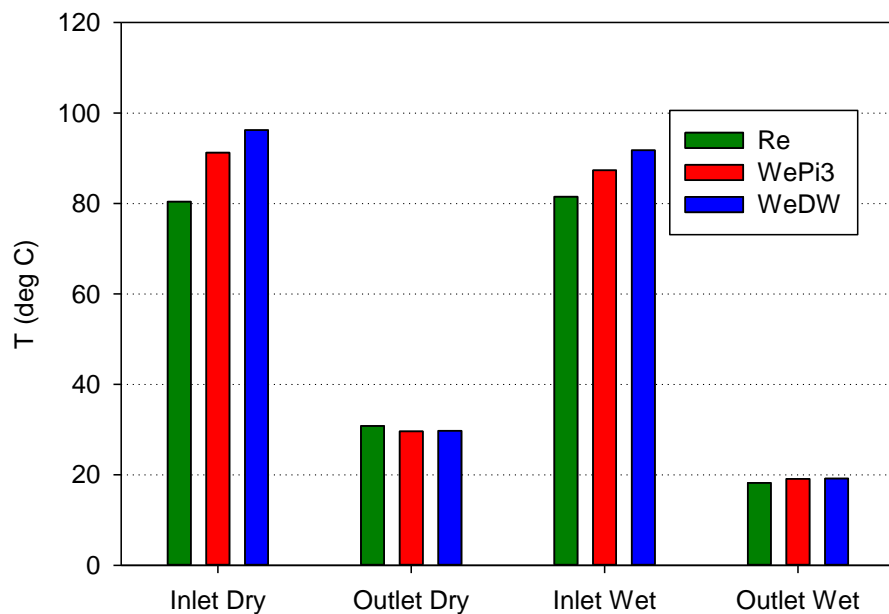
LE Surface Temperature



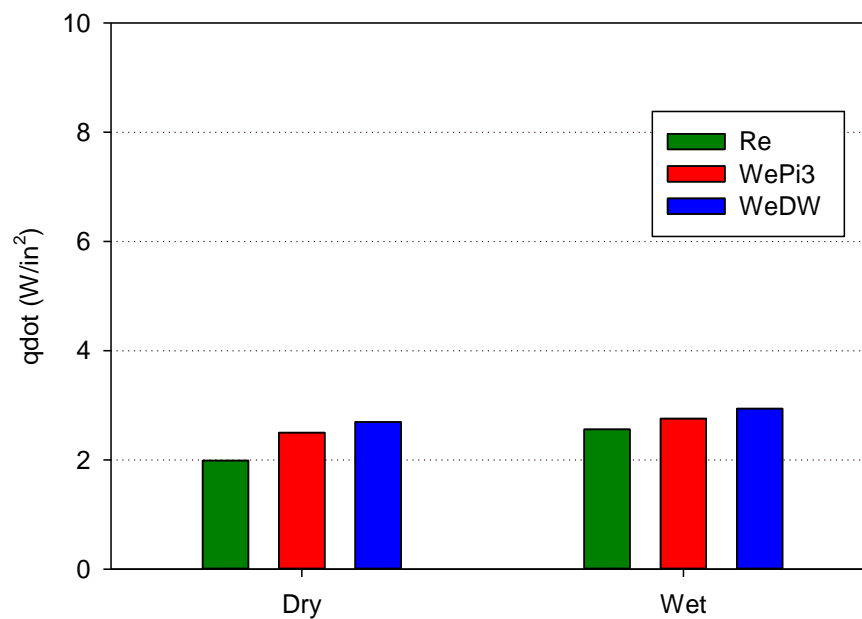


Warm Hold Scenario

Heated Air Temp.



Heated Air Energy Input





Conclusion

- Test conducted at NASA Icing Research Tunnel to evaluate new altitude scaling methods for thermal ice protection systems.
- Two Weber number-based scaled methods developed during a series of joint NASA and NRCC tests at AIWT.
- Results from IRT generally agreed with and supported the results from previous tests in NRCC.
- We-based scaling methods resulted in smaller ice accretion that formed farther upstream than the Re-based scaling methods.
- Additional tests required in altitude capable tunnels using full-scale models to better define the limits of physical relationships used to develop these scaling methods.

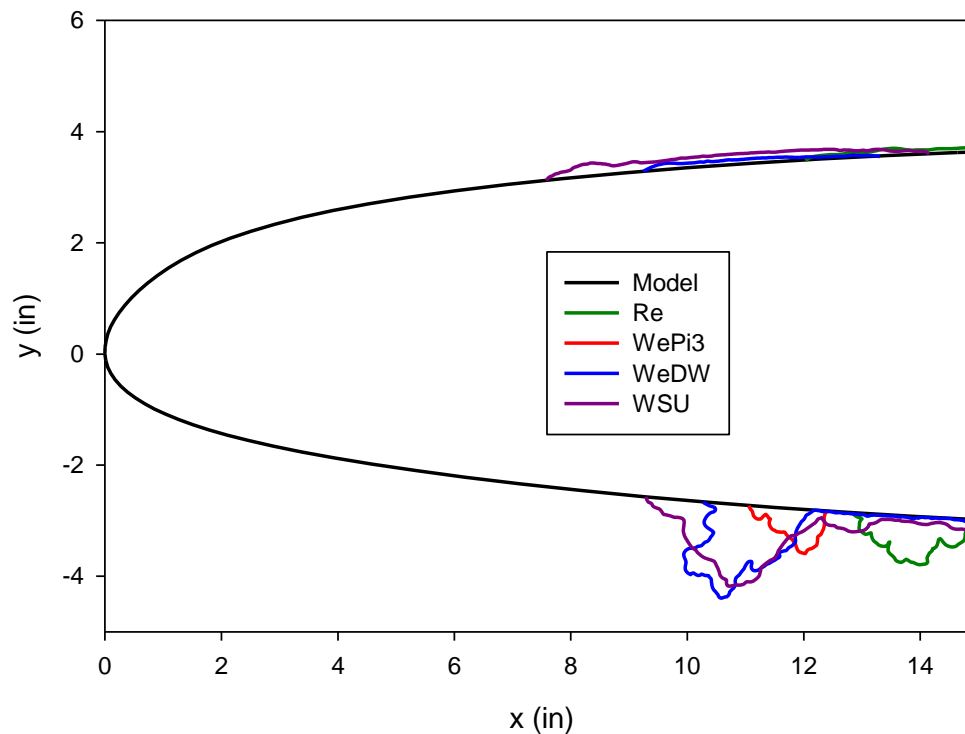


Extra Slides



WSU Warm Hold Scenario

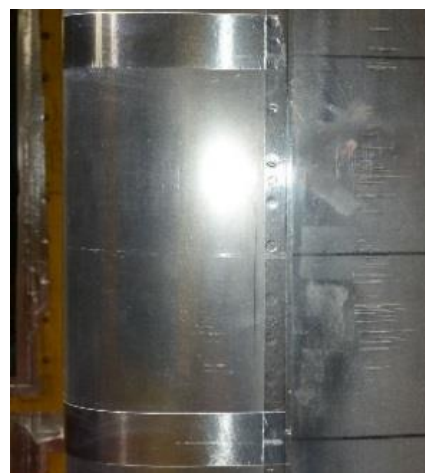
| Scale Method | Alt (ft) | V (kts) | T_s (deg C) | LWC (g/m^3) | MVD (μm) | m_w ($\text{g}/\text{m}^2\text{s}$) | Re_{2r} ($\times 10^6$) | WeDA | Pi3 | WeDW ($\times 10^6$) | Ice Mass (g) |
|------------------|----------|---------|---------------|-------------------------------|-----------------------|---|------------------------------------|------|------|------------------------|--------------|
| <i>Reference</i> | 15000 | 205 | -9.4 | 0.5 | 20 | 31.1 | 0.205 | 6065 | 1.73 | 8.04 | <i>N/A</i> |
| Re | 1312 | 126 | -6.3 | 0.82 | 27.2 | 31.1 | 0.205 | 3769 | 2.98 | 3.01 | 236.5 |
| WePi3 | 1835 | 164 | -7.6 | 0.43 | 24 | 21.0 | 0.264 | 6065 | 1.73 | 5.13 | 68.9 |
| WeDW | 2446 | 205 | -9.2 | 0.5 | 22.4 | 31.1 | 0.324 | 9715 | 2.23 | 8.04 | 266.2 |
| WSU | 1191 | 115 | -9.4 | 0.87 | 29 | 30.8 | 0.192 | 3229 | 3.03 | 2.53 | 483.3 |



WSU Warm Hold Scenario



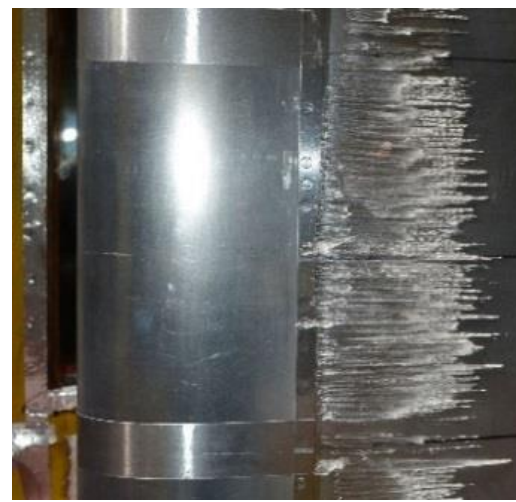
Re Scaling



WePi3 Scaling



WeDW Scaling

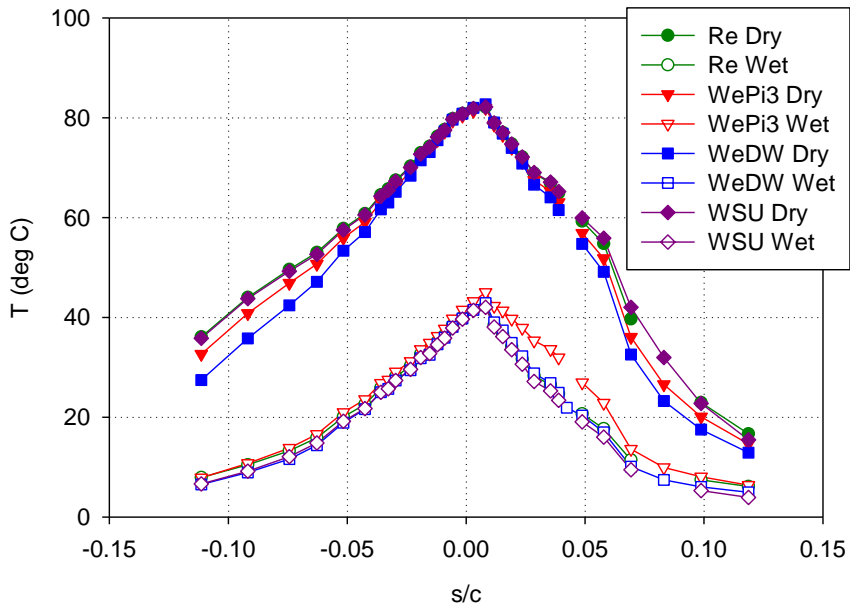


WSU Scaling

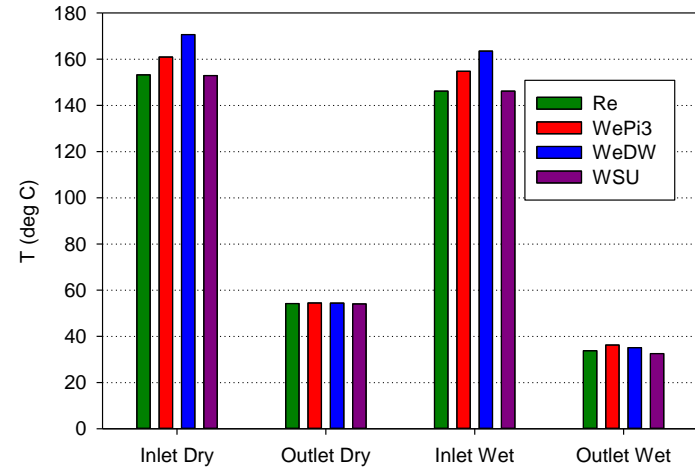


WSU Warm Hold Scenario

LE Surface Temp.



Heated Air Temp.



Heated Air Energy Input

