

# Creep, fatigue and environmental interactions and their effect on crack growth in superalloys

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## Presentation Overview

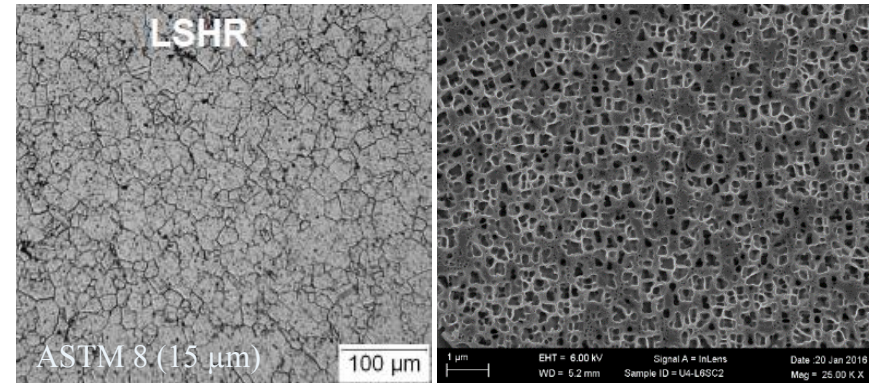
- **Complex interactions of creep/fatigue/environment control dwell fatigue crack growth (DFCG) in superalloys.**
- **Crack tip stress relaxation during dwells significantly changes the crack driving force and influence DFCG.**
- **Linear Elastic Fracture Mechanics,  $K_{max}$ , parameter unsuitable for correlating DFCG behavior due to extensive visco-plastic deformation.**
- **Magnitude of remaining crack tip axial stresses controls DFCG resistance due to the brittle-intergranular nature of the crack growth process.**
- **Proposed a new empirical parameter,  $K_{srf}$ , which incorporates visco-plastic evolution of the magnitude of remaining crack tip stresses.**
- **Previous work performed at 704°C, extend the work to 760°C.**

Material: Low Solvus High Refractory (LSHR) P/M nickel-base disk alloy

Wt. %	Al	B	C	Co	Cr	Mo	Ni	Nb	Ta	Ti	W	Zr
LSHR	3.5	.03	.045	20.4	12.3	2.7	Bal.	1.5	1.5	3.5	4.3	0.05

#### Four Supersolvus Heat Treatments Evaluated

Condition	Cooling Rate (°C/min)	Aging Treatment	Thermal Exposure
FC+2SA	202°C/min	855°C/4 h +775°C/8h	None
SC+2SA	72°C/min	855°C/4 h +775°C/8h	None
FC+2SA+440	202°C/min	855°C/4 h +775°C/8h	815°C-440 h
SC+2SA+440	72°C/min	855°C/4 h +775°C/8h	815°C-440 h



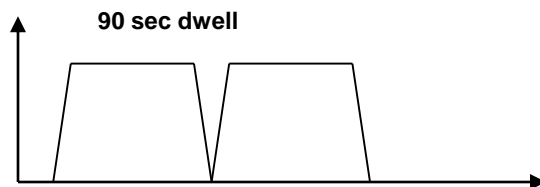
Testing performed at 704 °C and 760°C

#### Baseline FCG Testing:

- Cyclic FCG in Air and Vacuum; 0.333 to 30 Hz
- Dwell FCG in Air and Vacuum; 90 sec hold at  $\sigma_{\max}$
- Specimen Geometry: Surface Flaw (KB bar)

#### Baseline Stress Relaxation Testing:

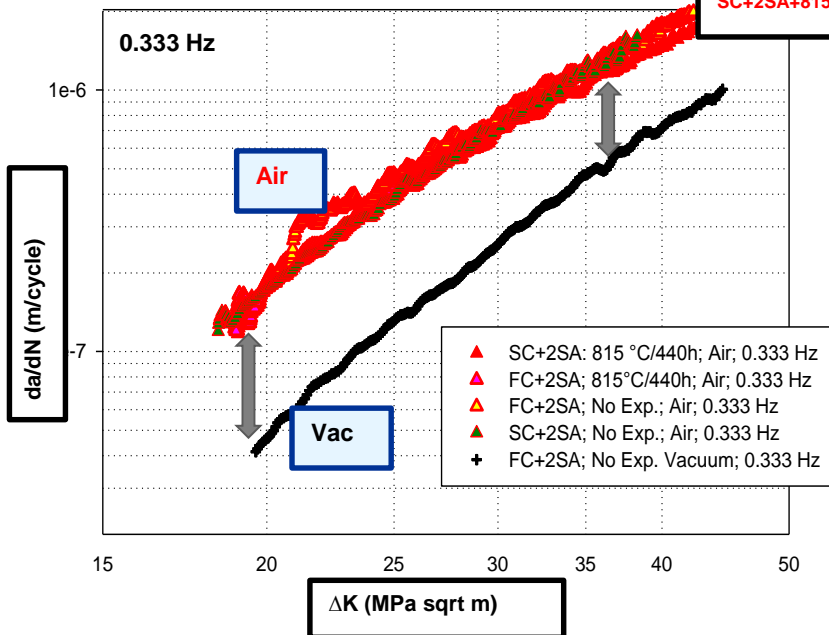
- Strained to 1% total strain
- Stress relaxation measured for 100 h.
- Specimen: Cylindrical (4.05 mm diam.)



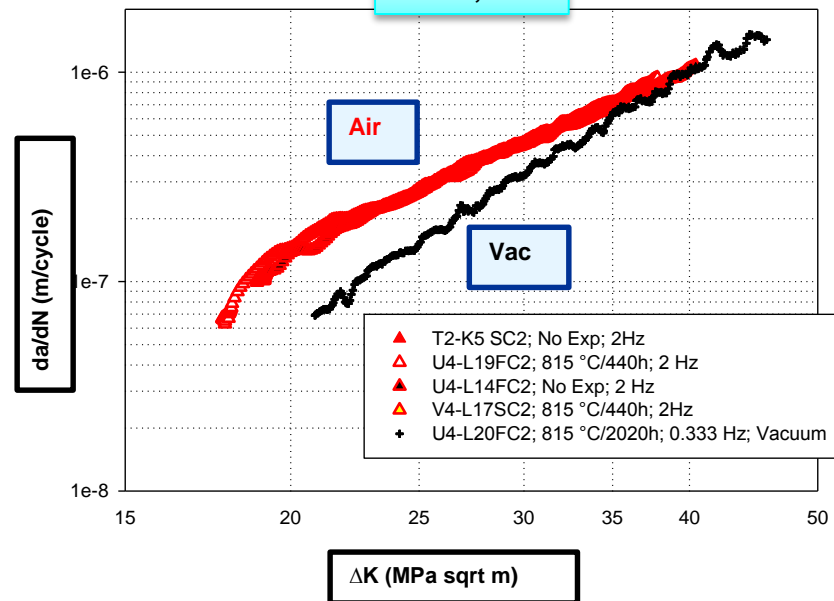
# Separating Environmental Effects from Stress Relaxation

704°C; 0.333 Hz

FC+2SA; SC+2SA  
 FC+2SA +815C@440h  
 SC+2SA+815C@440h

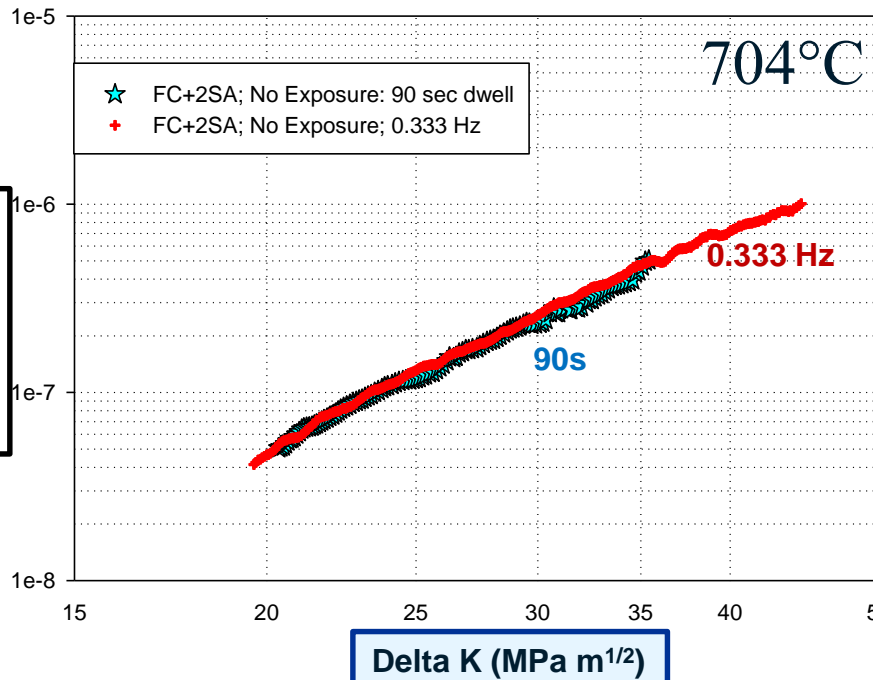


704°C; 2 Hz

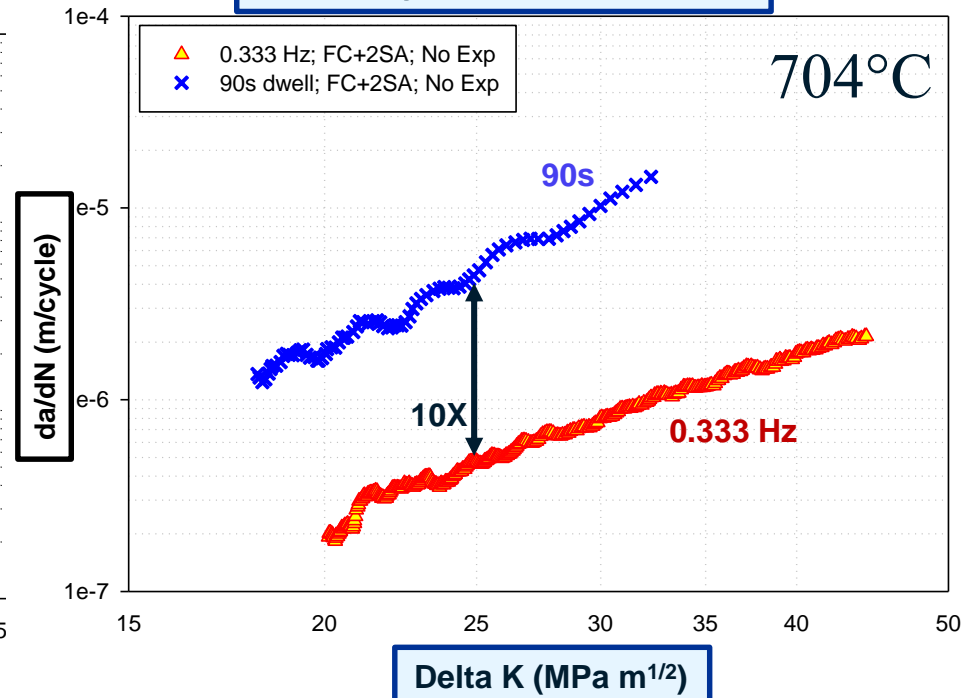


- All four conditions show faster cyclic FCGR in air than in vacuum – environmental debit.
- All conditions exhibited similar FCG resistance behavior but environmental effect is smaller at higher frequency.
- Assume these conditions possess similar *intrinsic* environmental resistance.
- Any differences in their dwell FCG resistance are then due to stress relaxation effects.

## Vacuum: Cyclic vs Dwell

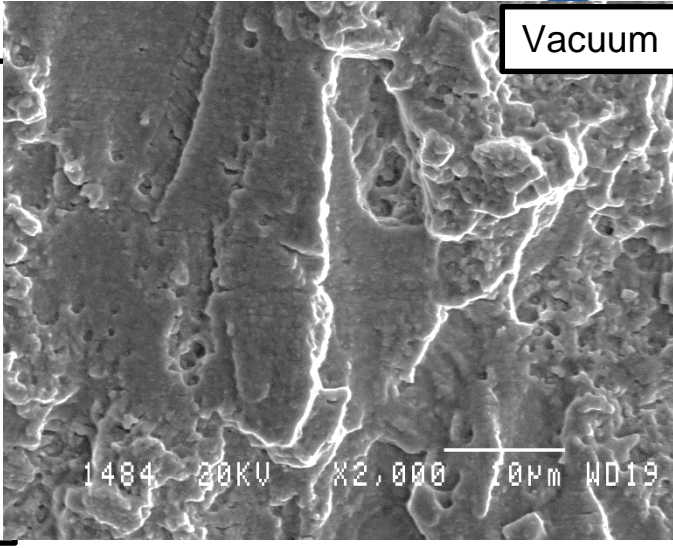
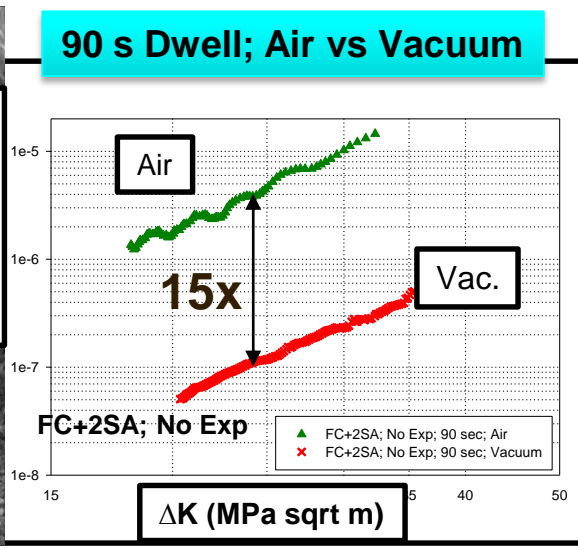
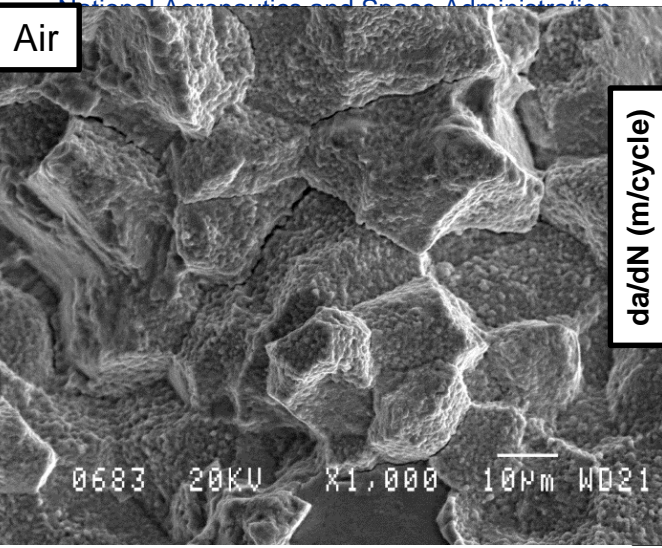


## Air: Cyclic vs Dwell



### At 704°C:

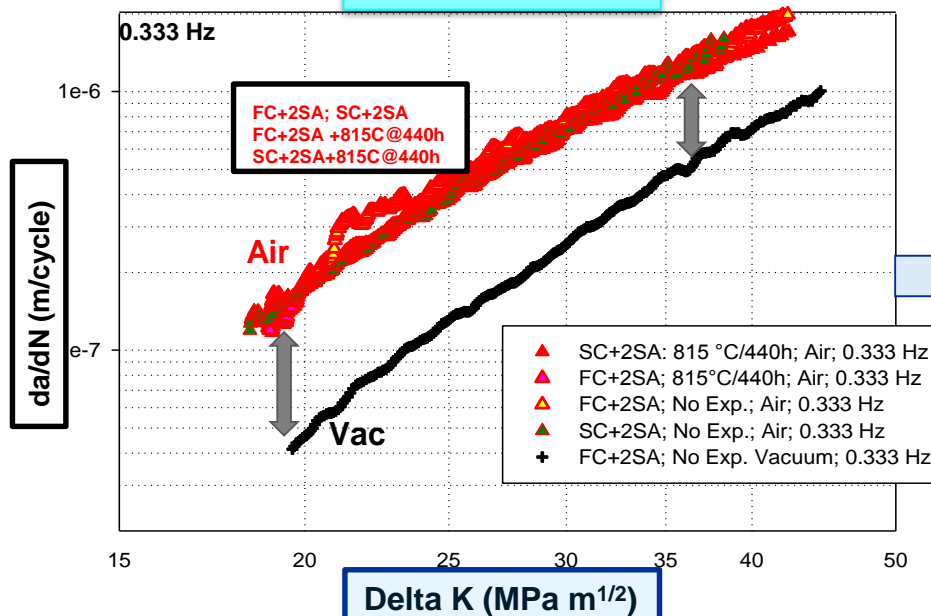
- 90 sec dwell FCG rates in vacuum same as cyclic FCG in vacuum – No Dwell Debit
- Creep crack growth does not contribute towards dwell crack growth
- An order of magnitude increase in DFCG in air due to environmental damage



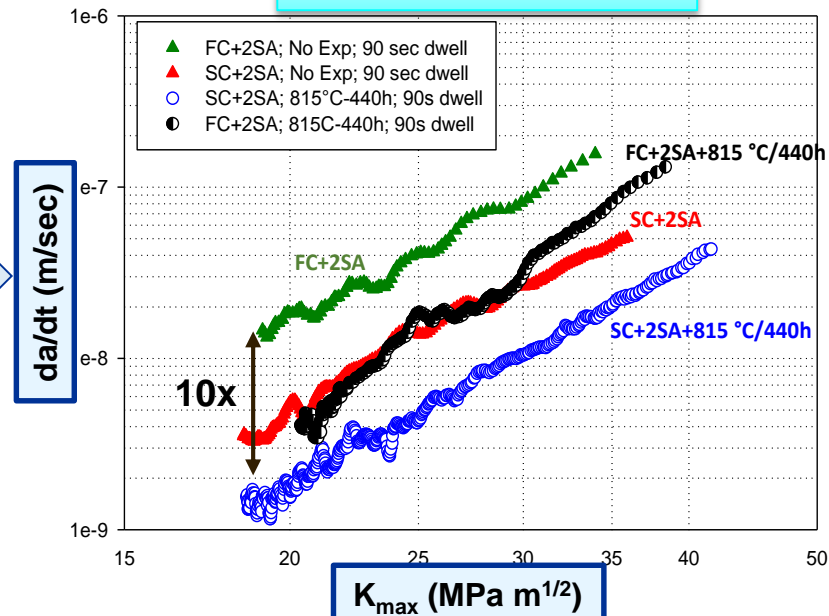
- Brittle-intergranular failure mode was operative in air for 90 sec dwell.
- Only transgranular failure mode operative in vacuum. No evidence of grain boundary sliding or microvoid coalescence found (classical creep crack growth **did not directly** contribute to DFCG).
- Grain boundaries are strong! Cracks avoid growth along grain boundaries when environmental embrittlement or creep mechanisms are not operative.



704°C; 0.333 Hz

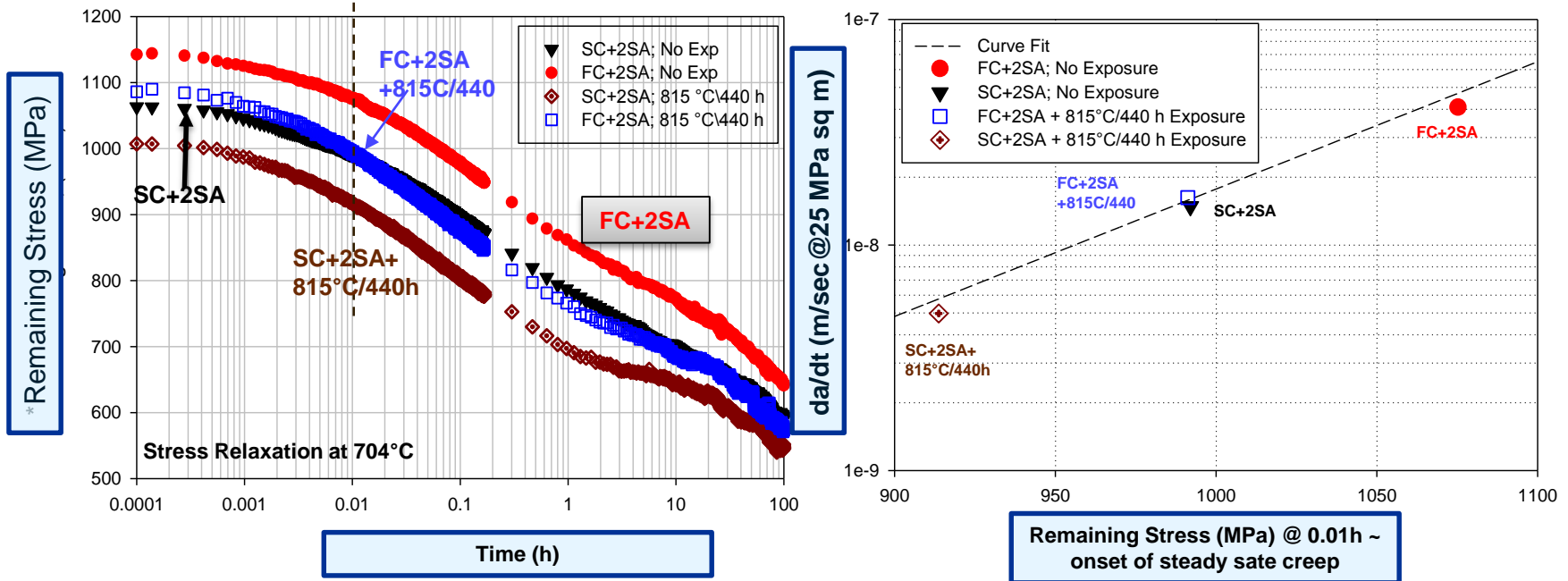


704°C 90s DFCG -Air



- Four heat treatments: similar env. resistance → 10x difference in DFCG.
- “Creepier heat treatments” i.e. slower cooling rates and thermal exposures improve DFCG resistance.
- Environmental resistance similar – DFCG differences due to stress relaxation.
- LEFM  $K_{max}$  parameter **unsuitable** for correlating visco-plastic influenced DFCG response.

## Relationship Between Stress Relaxation and DFCG



- Stress relaxation stresses decrease with slower cooling rates and  $\uparrow$  thermal exposure.
- Remaining stresses closely correlate with dwell fatigue crack growth  
Yet... Classical creep propagation mechanisms DO NOT contribute to crack growth
- Why is magnitude of remaining stresses important? What governs the relationship?

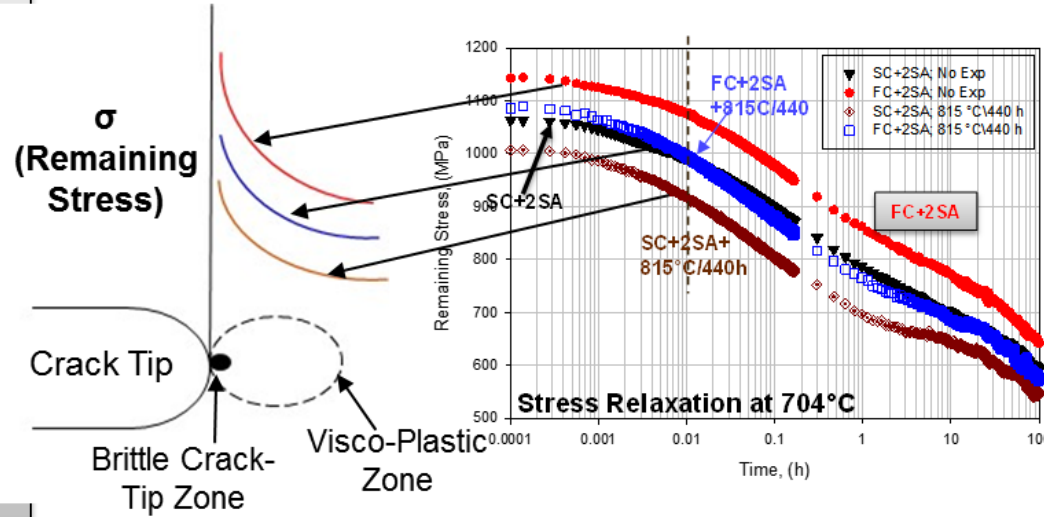
\*Remaining Stress = Relaxation Stress



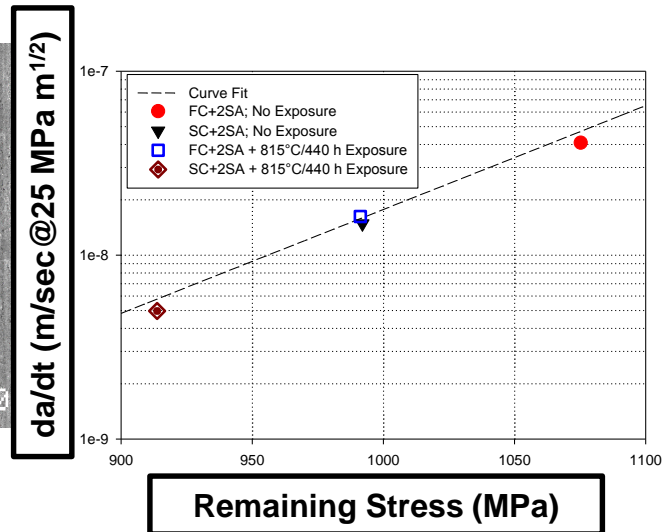
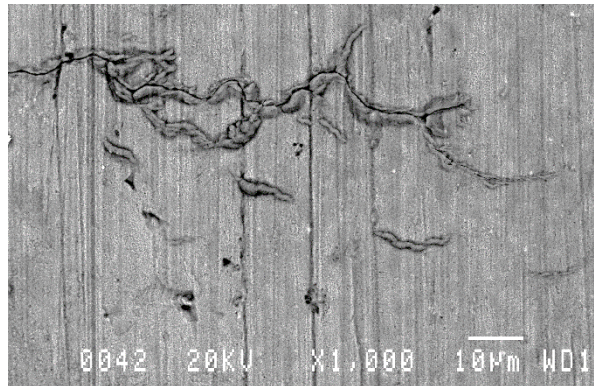
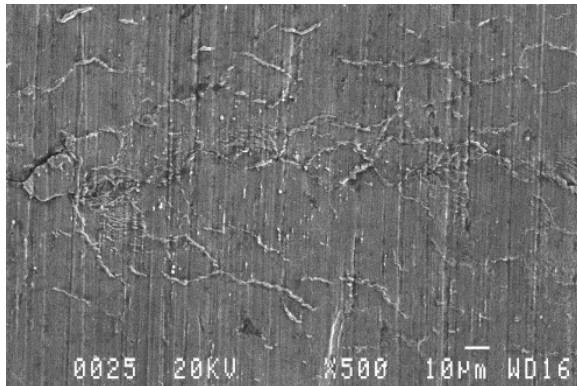


# DFCG Failure Mechanism

- Cracks grow through brittle-intergranular process controlled by crack tip tensile stress
- Magnitude of crack tip tensile stress controls DFCG propagation rates
- Stress relaxation behavior sets the magnitude of crack tip tensile stresses
- Strong, yet indirect relationship between stress relaxation and DFCG behavior



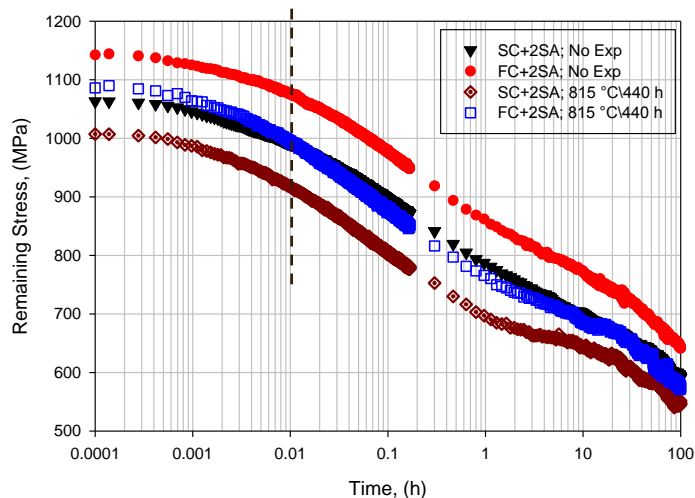
## Embrittled crack tip region – Interrupted 90s dwell tests





## New Empirical Parameter for Modeling Dwell Crack Growth in Air

Approach: Use stress relaxation results to simulate and normalize the differences in the crack tip tensile stresses under visco-plastic conditions



$$K_{srf} = K_{max} / SRF$$

$K_{srf}$  – modified stress intensity factor normalized by  $SRF$

$K_{max}$  – Applied LEFM stress intensity factor during dwells

$$SRF = (\sigma_0 / \sigma_m)^4$$

$SRF$  = stress relaxation factor

$\sigma_0$  = remaining stress at the onset of steady state creep (highest remaining stress condition)

$\sigma_m$  = remaining stress for other conditions – onset of steady state creep

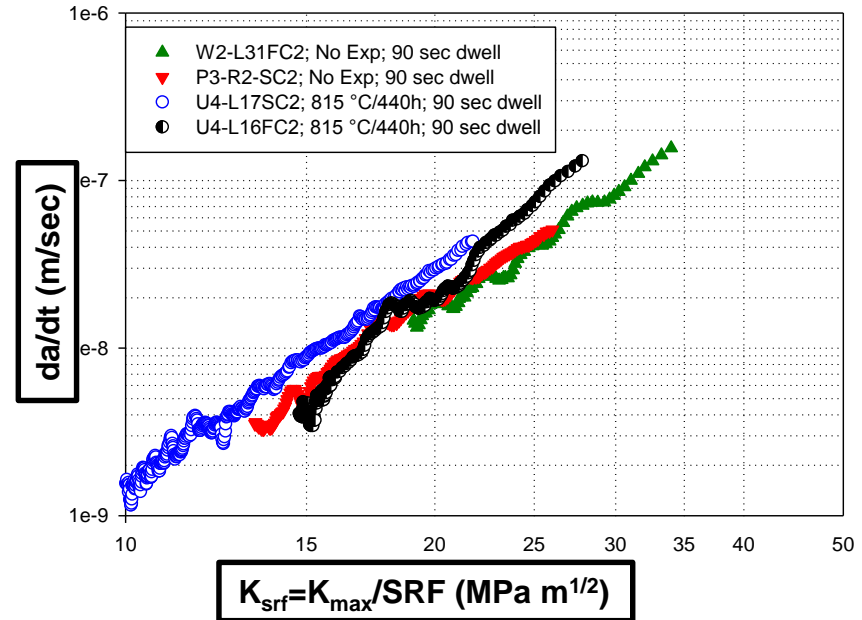
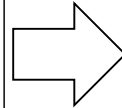
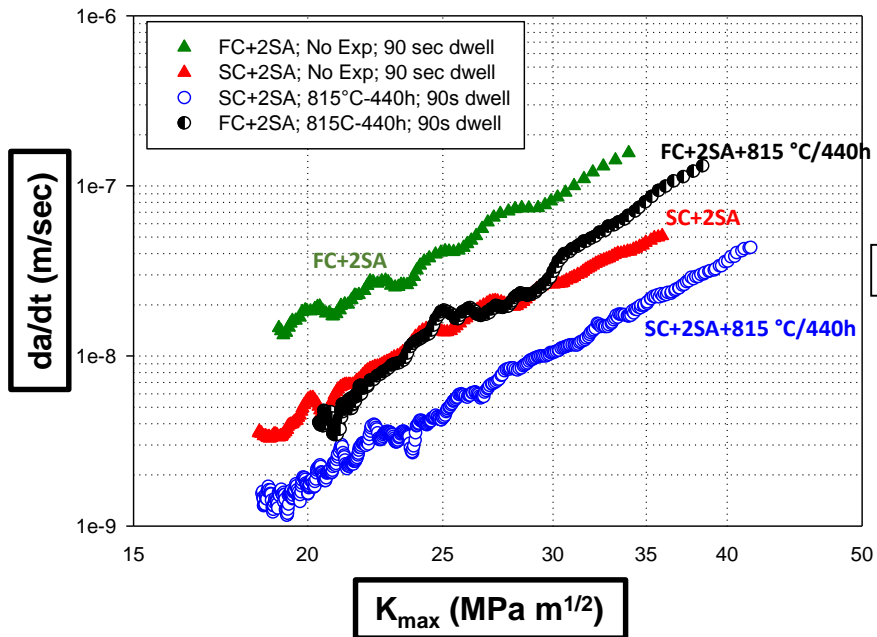
$$\dot{\epsilon} = A\sigma^{n1}t^m + B\sigma^{n2}$$

$n2 = 4$  (steady state creep component per the relaxation fit)



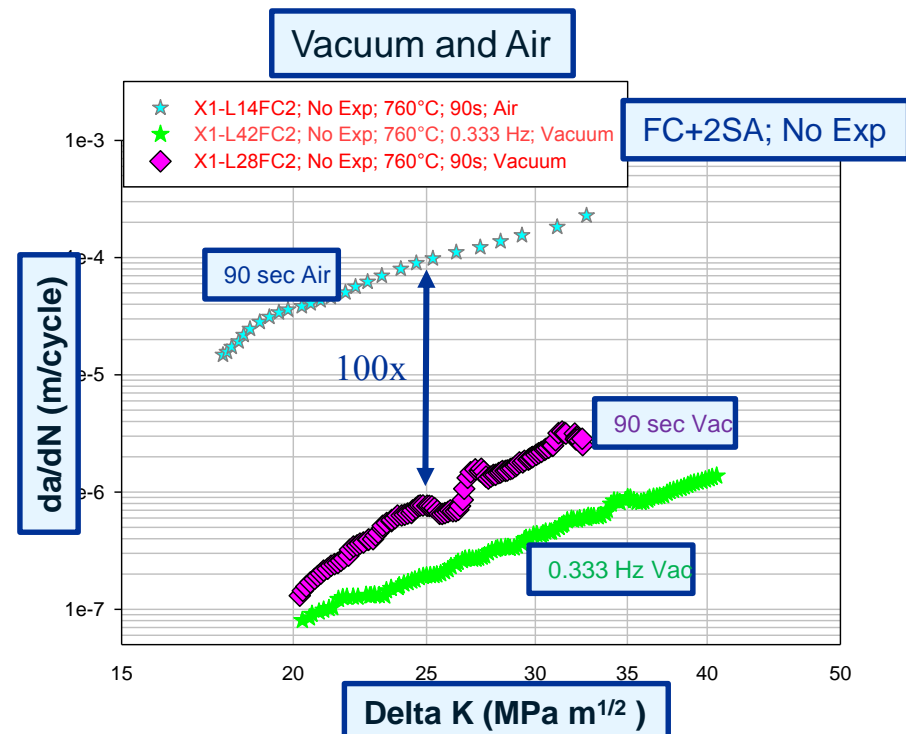
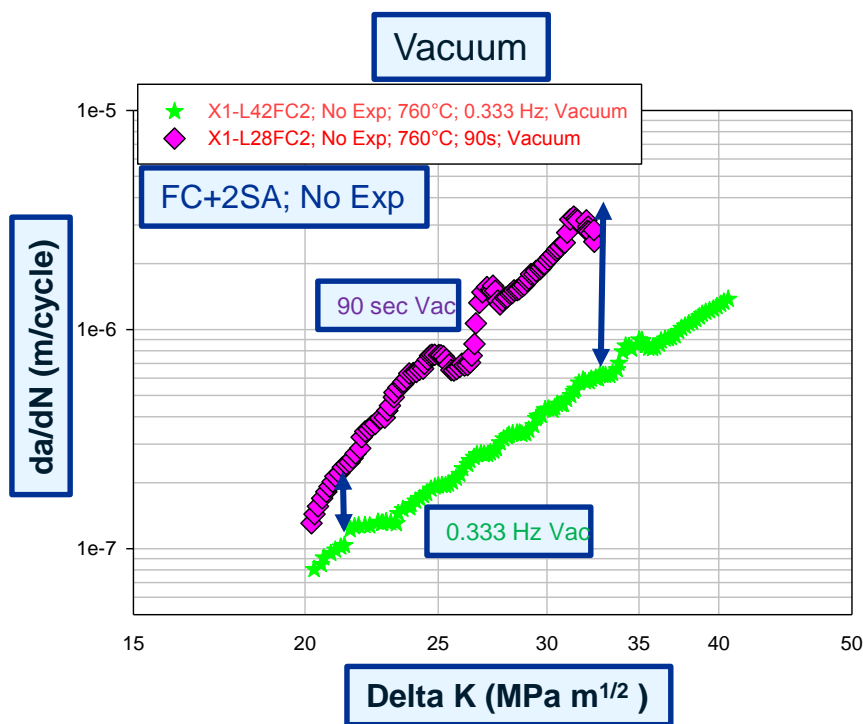
# New Empirical Parameter for Modeling Dwell Crack Growth in Air

## Plots of $da/dt$ vs $K_{max}$ and $K_{srf}$ : 704°C



- New  $K_{srf}$  parameter able to compensate for a 10x spread in DFCG rates using standard LEFM parameter.

## DFCG in Vacuum and Air – Behavior at 760°C



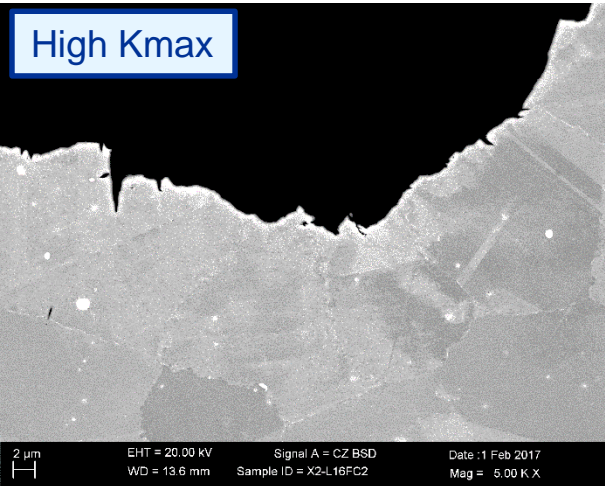
- In contrast to 704°C, creep crack growth occurs in vacuum at 760°C
- More significant at high  $\Delta K$  → increase in crack tip plasticity

- DFCG in air 100x faster than in vacuum.
- Environmental degradation is predominant... Ksrf approach may still be applicable.

# Comparison of Microstructural Damage Mechanisms

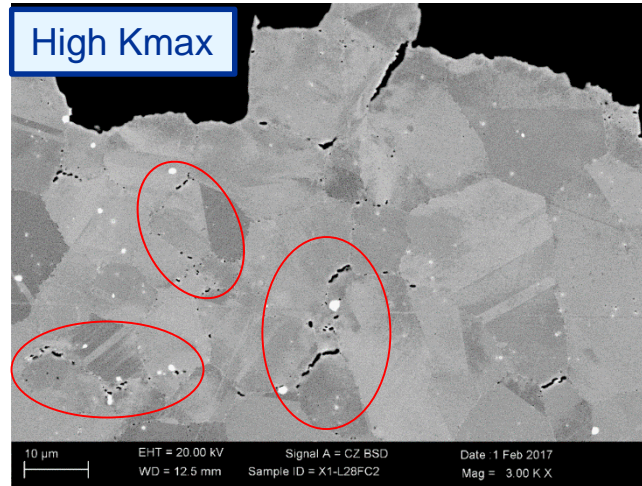
704°C – 90 sec; Vacuum

High Kmax



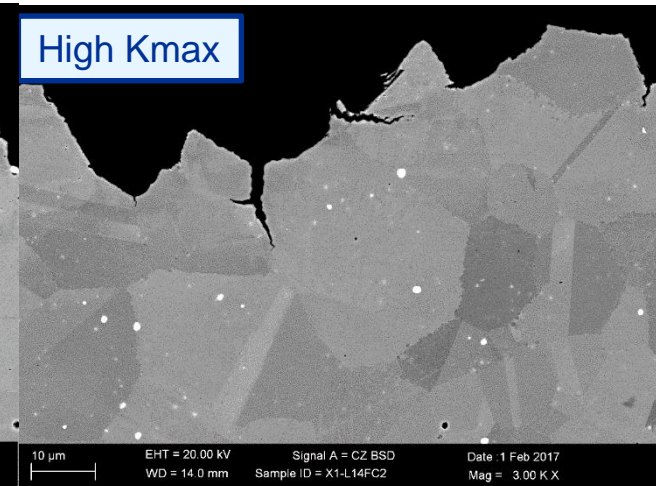
760°C – 90 sec; Vacuum

High Kmax



760°C – 90 sec; Air

High Kmax

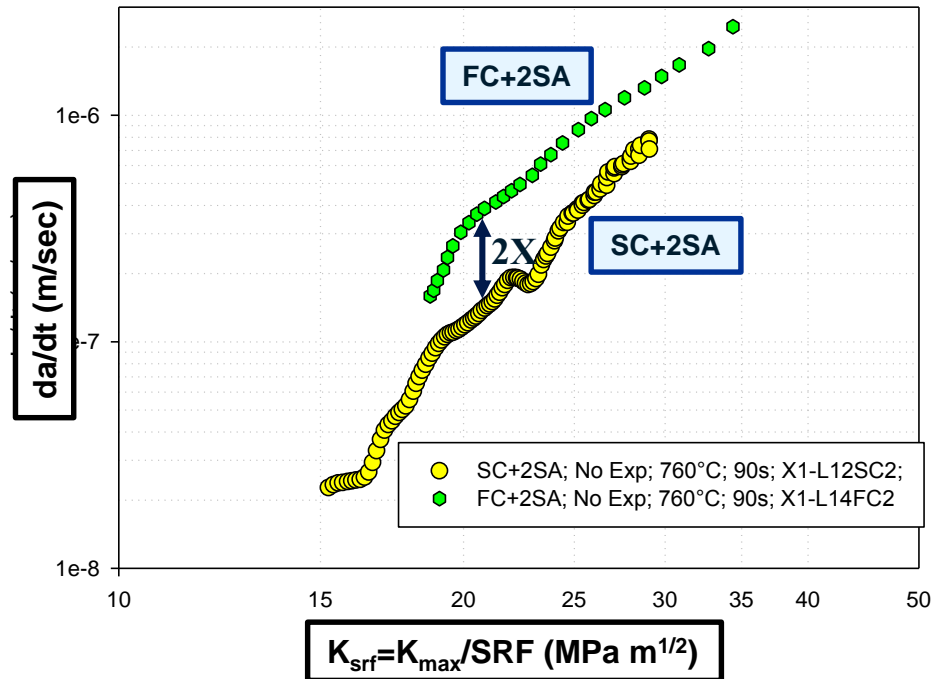
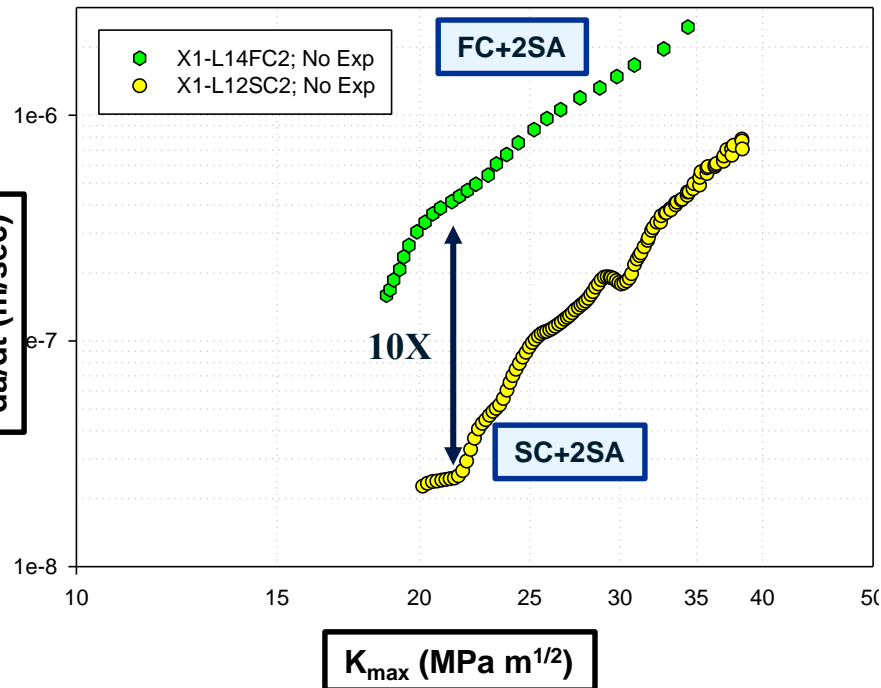


- Lower T. ; DFCG=Cyc FCG;
- No evidence of creep damage
- Transgranular failure mode

- High Temp; DFCG > CFCG
- Slow DFCG
- Grain boundary creep cavitation

- High T; DFCG >> CFCG
- Fast DFCG
- Environmentally induced intergr. failure mode dominant
- Inadequate time for creep cavitation

## 760°C DFCG; $K_{\max}$ vs $K_{\text{srf}}$



- Identical methodology used to calculate  $K_{\text{srf}}$  as at 704°C
- $K_{\text{srf}}$  correlated DFCG within 2X for FC+2SA and SC+2SA (no exposures)
- Other two conditions experienced likely specimen mixup at the vendor... No agreement between stress relaxation repeats, sorting out the issues...



## Conclusions

- **A new empirical parameter,  $K_{srf}$ , proposed to correlate DFCG in superalloys.**
- **The new parameter modifies  $LEFM\_K_{max}$  parameter by accounting for differences in visco-plastic evolution of the magnitude of remaining crack tip axial stresses.**
- **Magnitude of remaining crack tip axial stresses controls DFCG resistance due to the brittle-intergranular nature of the crack growth process.**
- **The parameter works well at  $704^{\circ}C$  and looks promising even at  $760^{\circ}C$ .**
- **Creep crack growth mechanisms are active at  $760^{\circ}C$  but are still considerably lower than the environmentally induced DFCG debit.**



# Acknowledgment

Special acknowledgment to Mr. Andrew Ring for his excellence and diligence in performing a multitude of complex mechanical testing.