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ORIGINAL ARTICLE

Optimum design of B-series marine propellers

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KEYWORDS

Optimization; Single objective; B-Series propeller **Abstract** The choice of an optimum marine propeller is one of the most important problems in naval architecture. This problem can be handled using the propeller series diagrams or regression polynomials. This paper introduces a procedure to find out the optimum characteristics of B-series marine propellers. The propeller design process is performed as a single objective function subjected to constraints imposed by cavitation, material strength and required propeller thrust. Although optimization software of commercial type can be adopted to solve the problem, the computer program that has been specially developed for this task may be more useful for its flexibility and possibility to be incorporated, as a subroutine, with the complex ship design process.

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1. Introduction

Naval architects can easily design an optimized propeller with the theoretical propeller design methods (lifting-line/surface theories) using a computer without the geometry constraints seen in series propellers [1]. However, series propellers are still valuable; they are still widely used in the preliminary design of light or moderately loaded propellers [2]. Moreover, for those

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who cannot afford lifting surface software, traditional series propellers are good choices. Among the propeller series, the B-series is one of the commonly used series.

In this study, a computer program has been specially developed to find the optimum characteristics of any B-series propeller. The propeller design process is handled as a single objective function subjected to several constraints such as cavitation, material strength and the required propeller thrust.

2. Optimization problem

The main difficulty in most optimization problems does not lie in the mathematics or methods involved. It lies in formulating the objective all the constraints.

The propeller design problem has been handled as a multiobjective constrained optimization problem [1–4]. There are two principal ways to handle 'multi-objective' problems, both leading to single objective optimization problems [5]:

• one objective is selected and the other objectives are formulated as constraints.

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Nomenclature

A _E	propeller expanded area m^2	si	exponent of (1) in $K_{\rm T}$ equation
A	propeller disk area m ²	si'	exponent of (I) in K_0 equation
C_{\circ}	r_{rec} blade chord length at 0.75 <i>R</i> m	51 T	propeller thrust N
C_0	regression coefficient of torque coefficient	Т	calculated propeller thrust N
$C_{\rm m}$	regression coefficient of thrust coefficient	T_{Cal}	required propeller thrust. N
	regression coefficient of thrust coefficient		required propener tinust, in
D	propeller diameter, m	1	propeller blade thickness, m
$H_{\rm F}$	AP height of propeller aperture, m	$t_{\rm d}$	thrust deduction
J	advance coefficient	ti	exponent of (P/D) in $K_{\rm T}$ equation
$K_{\rm Q}$	torque coefficient	ti'	exponent of (P/D) in K_Q equation
K _T	thrust coefficient	t _{min}	minimum thickness of propeller blade, m
N	propeller rotating speed, rpm $N = 60 * n$	ui	exponent of $A_{\rm E}/A_{\rm O}$ in $K_{\rm T}$ equation
$N_{\rm P}$	number of propellers	ui′	exponent of $A_{\rm E}/A_{\rm O}$ in $K_{\rm Q}$ equation
п	propeller rotating speed, rps	$V_{\rm A}$	speed of advance, m/sec
M	number of the design variables	V_{S}	ship speed, m/sec
Р	propeller blade pitch, m	Vi	exponent of Z in $K_{\rm T}$ equation
$P_{\rm C}$	_L propeller immersion, m	Vi′	exponent of Z in $K_{\rm O}$ equation
$P_{\rm D}$	developed power, kW	W	wake fraction
$P_{\rm S}$	shaft power per blade, kW	Ζ	number of propeller blades
$P_{\rm O}$	static pressure at the centerline of the propeller	$\Delta K_{\rm O}$	correction of torque coefficient
	shaft, Pa	$\Delta K_{\rm T}$	correction of thrust coefficient
$P_{\rm V}$	vapor pressure, Pa	η	propeller efficiency
Р	total number of constraints	$\eta_{\rm max}$	maximum propeller efficiency
q	number of inequality constraints	ρ	water density, kg_m/m^3
Rn	Reynolds number	v	water kinematic viscosity, m^2/s
R _T	ship total resistance, N		.,
Sc	maximum allowable stress of the propeller		
- 0	material, MPa		

• a weighted sum of all objectives forms the optimization objective function.

The rather arbitrary choice of weight factors makes the optimization model obscure and the first option is mostly preferred [5]. In this study, the propeller design problem is handled as a constrained optimization problem according to the first option.

Any constrained optimization problem can be handled according to the optimization model shown in Fig. 1.

This optimization problem can be formulated as follows [6]:

Find
$$X = \begin{cases} x_1 \\ x_2 \\ \vdots \\ x_m \end{cases}$$
 (1)

which maximizes an objective function called f(X) subjected to the following constraints:

$$g_{j}(X) \leq 0, \quad j = 1, 2, \dots, q$$

ad
$$\ell_{j}(X) = 0, \quad j = q + 1, \ q + 2, \dots, p$$
(2)

and

where, $g_j(X)$ and $\ell_j(X)$ are the inequality and the equality constraints, respectively.

3. B-series propeller

B-series propellers were developed in the Netherlands Ship Model Basin, and the section of the blade was improved later.



Figure 1 Optimization model for constrained problems.

For any B-series propeller, the thrust and torque coefficients can be expressed as functions of the blade number (Z), blade

area ratio (A_E/A_O) , pitch ratio (P/D), and advance coefficient (*J*) as follows:

$$K_{\rm T} = \sum_{i=1}^{39} C_{\rm si} J^{\rm si} \left(\frac{P}{D}\right)^{\rm ti} \left(\frac{A_{\rm E}}{A_{\rm O}}\right)^{\rm ui} Z^{\rm vi}$$
(3)

and,

$$K_{\mathcal{Q}} = \sum_{i=1}^{47} C_{\mathrm{mi}} \, \mathcal{J}^{\mathrm{s}\mathrm{i}^{\prime}} \left(\frac{P}{D}\right)^{\mathrm{t}\mathrm{i}^{\prime}} \, \left(\frac{A_E}{A_O}\right)^{\mathrm{u}\mathrm{i}^{\prime}} \, Z^{\mathrm{v}\mathrm{i}^{\prime}} \tag{4}$$

where, C_{si} and C_{mi} are the regression coefficients of the thrust and torque coefficients, respectively. The values of the coefficients and exponents involved in Eqs. (3) and (4) are given in [7].

If Reynolds number of a propeller at 0.75R is greater than $2 * 10^6$, corrections to the thrust and torque coefficients must be taken into consideration [1].

$$(\mathbf{Rn})_{0.75R} = \frac{C_{0.75R} * \sqrt{[V_{\rm A}^2 + (0.75\pi \ n \ D)^2]}}{\nu}$$
(5)

where, $C_{0.75R}$ is the blade chord length at 0.75R and V_A is the advance velocity (m/s).

3.1. Objective function

The most common objective function, for the optimum marine propellers, is the propeller maximum efficiency (η_{max}) [8]. The efficiency of marine propellers can be computed as follows:

$$\eta = \frac{J}{2\pi} \cdot \frac{K_{\rm T}}{K_{\rm Q}} \tag{6}$$

3.2. Constraints

3.2.1. Cavitation constraint

Cavitation could affect a propeller's performance and needs to be considered in the propeller design process. A simple way to avoid cavitation is to increase blade area ratio. The minimum blade area ratio required to avoid cavitation was suggested by Keller [1] as follows:

$$\left[\frac{A_{\rm E}}{A_{\rm O}}\right]_{\rm min} = \frac{(1.3+0.3Z)T}{(P_{\rm O}-P_{\rm V})D^2} + K \tag{7}$$

where, $(A_{\rm E}/A_{\rm O})_{\rm min}$ is the minimum expanded area ratio. The coefficient *K* equals 0.1 for twin-screws ships, and 0.2 for single-screw ships.

3.2.2. Strength constraint

To achieve adequate blade thickness and thus ensure material strength, the following formula can be used to determine the minimum ratio of blade thickness at 0.7R to the diameter [4]:

$$\left[\frac{t_{\min}}{D}\right]_{0.7R} = 0.0028 + 0.21\sqrt[3]{\frac{[3183.87 - 1508.15(P/D)]P_{\rm S}}{1266652.04nD^3(S_{\rm C} + 20.9D^2n^2)}}$$
(8)

where, $(t_{\min}/D)_{0.7R}$ is the minimum propeller blade thickness ratio at 0.7R.

According to the B-series propeller geometry [7], the ratio of maximum thickness of propeller blade at each section to the propeller diameter is given as shown in Table 1.

 Table 1
 Maximum blade thickness % of D for B-series propellers.

r/R	Max. blade t	Max. blade thickness (% of D)			
	Z = 3	Z = 4	Z = 5		
0.2	4.06	3.66	3.26		
0.3	3.59	3.24	2.89		
0.4	3.12	2.82	2.52		
0.5	2.65	2.40	2.15		
0.6	2.18	1.98	1.78		
0.7	1.71	1.56	1.41		
0.8	1.24	1.14	1.04		
0.9	0.77	0.72	0.67		
1.0	0.30	0.30	0.30		

As shown in table 1, the ratio of maximum thickness of propeller blade at 0.7R to the diameter is given as follows:

$$\begin{bmatrix} t \\ \overline{D} \end{bmatrix}_{0.7R} = \begin{cases} 0.0171 & \text{for } Z = 3 \\ 0.0156 & \text{for } Z = 4 \\ 0.0141 & \text{for } Z = 5 \end{cases}$$
(9)

Using Eq. (8) and the geometry of B-series propellers, the required propeller blade thickness can be obtained as follows:

$$\left[\frac{t}{D}\right]_{0.7R} \geqslant \left[\frac{t_{\min}}{D}\right]_{0.7R} \tag{10}$$

3.2.3. Thrust constraint

The calculated propeller thrust (T_{Cal}) must be equal to the required thrust (T_{R}). The propeller thrust can be calculated as follows:

$$T_{\rm CAL} = K_{\rm T} \ \rho \ n^2 \ D^4 \tag{11}$$

$$T_{\rm R} = \frac{R_{\rm T}}{N_{\rm P} * (1 - t_{\rm d})} \tag{12}$$

where, $R_{\rm T}$ is the ship total resistance, $N_{\rm P}$ is the number of propellers and $t_{\rm d}$ is the thrust deduction.

4. Developed computer program

A computer program has been specially developed to find the optimum characteristics of any B-series propellers. This program starts with an initial feasible point and proceeds towards the optimum point according to the steps shown in Fig. 1. The detailed procedure of this program is summarized in the flow chart shown in Fig. 2.

5. Case study

The problem is to find the optimum characteristics $(D, A_E/A_O, P/D, J)$ of four bladed B-type propeller of a twin screw ship. The propeller diameter (D) is restricted at specified value to suit the ship lines and to provide acceptable hull clearances to avoid hull vibration. Ship speed (V_S) and corresponding resistance (R_T) are given for this design condition. Wake fraction (w) and thrust deduction (t_d) are given for this design condition. This problem can be formulated as follows:



Figure 2 Flow chart for B-series marine propellers design process.

Table 2 Input data.		
Ship total resistance $(R_{\rm T})$		
hip speed $(V_{\rm S})$		
Number of propellers $(N_{\rm P})$		2
Number of propeller blades (Z)		4
Height of propeller aperture (H_{PAP})		1.25
Propeller immersion (P_{CL})		2.0
Wake fraction (w)		0.20
Thrust deduction (t_d)		0.15
Max. allowable stress of propeller material $(S_{\rm C})$		260
Boundary constraint		
Design variable	Lower Limit	Upper Limit
Propeller diameter (D)	0.25	$0.8 * H_{PAP}$
Area ratio $(A_{\rm E}/A_{\rm O})$	0.35	0.85
Pitch ratio (P/D)	0.7	1.5
Advance coeff. (J)	0.3	1.5



Figure 3 Output results of the developed computer program.

Use the input data which are shown in Table 2 to find $(D, A_{\rm E}/A_{\rm O}, P/D, J)$ which maximize the propeller efficiency (η) when subjected to the following constraints:

- Cavitation constraint: the expanded area ratio should be larger than a minimum value in order to avoid cavitation, $(A_{\rm E}/A_{\rm O}) \ge (A_{\rm E}/A_{\rm O})_{\rm min}$ where $(A_{\rm E}/A_{\rm O})_{\rm min}$ can be calculated using Eq. (7).
- Thrust constraint: the calculated propeller thrust (T_{cal}) has to match the design requirement, as follows: $T_{cal} = T_{required}$.
- *Strength constraint:* to ensure adequate material strength, a minimum propeller blade thickness is required, as shown in Eq. (8).

 $\left[\frac{t}{D}\right]_{0.7R} > \left[\frac{t_{\min}}{D}\right]_{0.7R}$

• *Boundary constraint:* to obtain an optimum design point, the design variables must lie in an acceptable domain.

In this part, the problem under consideration is carried out in details by using the following techniques:



Figure 4 Open water diagram.

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Objective value:		0.6201316
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Variable	varue	Reduced Cost
[D]	1.0000000	0.000000
[AE/Ao]	0.5445637	0.000000
[J]	0.7841496	0.000000
[P/D]	1.0820710	0.00000
(n)	10 496980	0,000000
	0.0367040	0.000000
[K _T]	0.1822916	0.000000
[TCAL]	20588.24	0.000000
[P _D]	546.54810	0.000000
[Pp per blade]	68.318520	0.000000
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Figure 5 Output results of Lingo software.

• Using the developed computer program.

• Using a commercial software (Lingo).

Table 3Output results.

Items	Developed program	Lingo
Objective: $\eta_{\text{max.}}$	0.620	0.62013
Design variables		
D	1.00	1.00
$A_{\rm E}/A_{\rm O}$	0.546	0.54456
P/D	1.075	1.082
J	0.78	0.784
[n] RPS	10.57	10.497
Coefficients		
KT	0.1809	0.1823
Ko	0.0362	0.0367
Power in kwatt		
P _D	546.66	546.548
$P_{\rm D}$ (Blade)	68.33	68.318

5.1. Using the developed computer program

The data which are shown in Table 2 are used to perform the problem under consideration by using the developed computer program. The output results of this program are shown in Fig. 3.

The obtained optimum point is plotted on the open water diagram for this design condition as shown in Fig. 4.

5.2. Using a commercial software (Lingo)

Lingo [9] is a comprehensive tool designed to make building and solving linear, nonlinear and integer optimization models. The problem under consideration is recalculated using LINGO software [9] and the output results are shown in Fig. 5.

5.3. Analysis of the results

The output results obtained by the presented solutions are shown in Table 3. It is clear that, the output results of Lingo software are found in a good agreement with those obtained by the developed program.

6. Conclusions

- The design process of a series propeller by traditional calculation or chart methods is a tedious job due to the multiple parameters and constraints involved.
- To handle the multi-objective constrained problem of the propeller design, one objective is selected and the other objectives are formulated as constraints.
- The developed computer program represents a tailored and simple tool to find the optimum characteristics of any B-series marine propeller. It is more flexible to use it as a subroutine in global ship design problems than to use commercial optimization

software.

References

- C. Jeng-Horng, S. Yu-Shan, Basic design of a series propeller with vibration consideration by genetic algorithm, J. Mar. Sci. Technol. 12 (2007) 119–129.
- [2] E. Benini, Multi-objective design optimization of B-screw series propellers using evolutionary algorithm design, J. Mar. Sci. Technol. 40 (2003) 229–238.
- [3] J.-B. Suen, J.-S. Kouj, Genetic algorithm for optimal series propeller design, in: Proceeding of the Third International Conference on Marine Technology, ORDA 99, Poland, 1999.
- [4] M.M. Karim, M. Ikehata, A genetic algorithm (GA)-based optimization technique for the design of marine propeller, in: Proceeding of the Propeller/Shafting Symposium, Virginia Beach, USA, 2000.
- [5] H. Schneekluth, V. Bertram, Ship Design for Efficiency and Economy, Butterworth, Heinemann, 1998.
- [6] S.S. Rao, Optimization Theory and Applications, second ed., Wiley Estern Limited, India, 1991.
- [7] M.W.C. Oosterveld, V. Oossanen, Further Computer Analyzed Data of The Wageningen B-series, I.S.P., vol. 23, July 1975.
- [8] M.M. Gaafary, Computerized method for propeller design of optimum diameter and rpm, in: 13th Congress of IMAM, Istanbul, Turkey, October 2009.
- [9] Lindo Systems Inc., Lingo Software, Version (9), < http:// www.lindo.com/>, April 2006.