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# Applied optimal shape design

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## Abstract

This paper is a short survey of optimal shape design (OSD) for fluids. OSD is an interesting field both mathematically and for industrial applications. Existence, sensitivity, correct discretization are important theoretical issues. Practical implementation issues for airplane designs are critical too.

The paper is also a summary of the material covered in our recent book, *Applied Optimal Shape Design*, Oxford University Press, 2001.

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## 1. Industrial demand

Since control is a natural desire once the simulation is completed, the applications of OSD are uncountable. For instance the design of a harbor which minimizes the waves coming from far can be done at little cost by standard optimization methods once the numerical simulation of Helmholtz equation is mastered (see Ref. [1]) as shown in Fig. 1. Other applications include

- weight reduction in car engine, aircraft structures, etc,
- electromagnetically optimal shapes, such as in stealth airplanes,
- wave canceling fore bulbe in boat design,
- drag reduction for airplanes, cars and boats.

However, there are no automatic solutions to these problems because most of the time engineering design is made of compromises due to the multi-disciplinary aspects of the problems (see Fig. 2),

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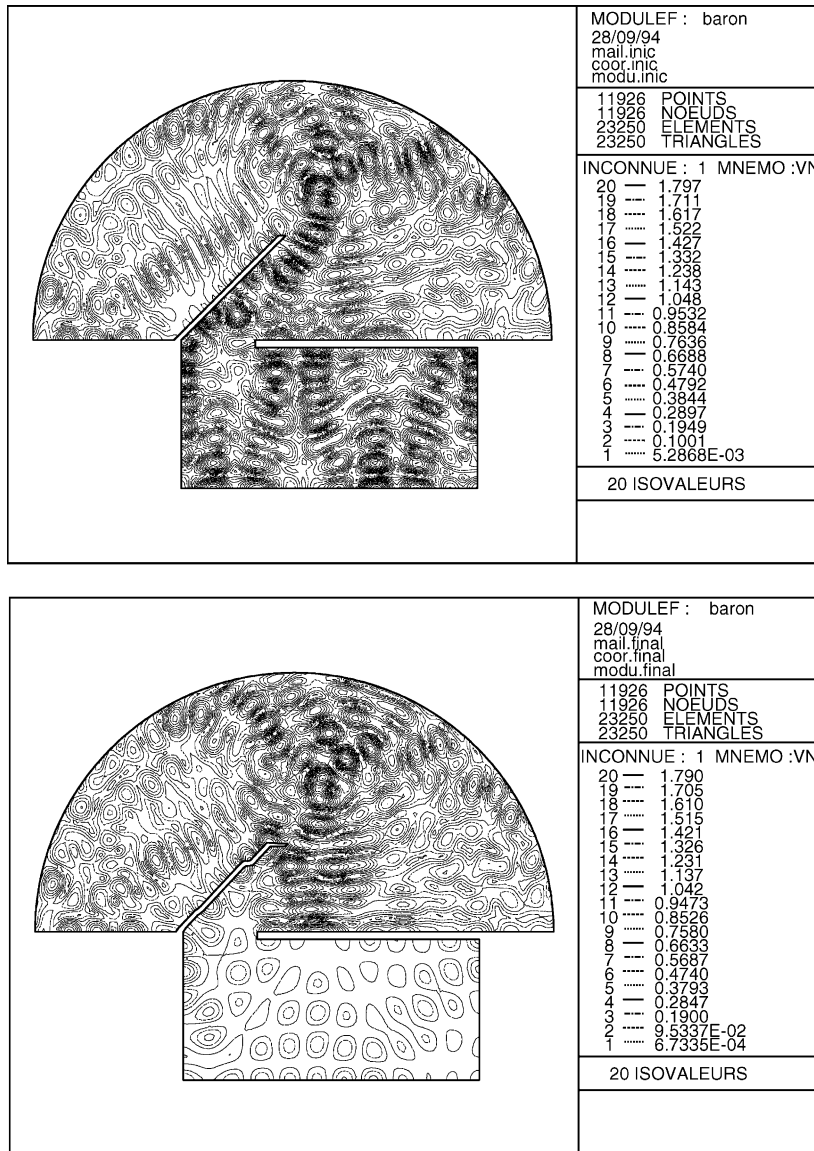


Fig. 1. Optimization of a brake-water (computed by A. Baron).

the necessity of doing *multi-point constrained design* and because the solvers are not always made in house and appear as *black-box* solvers.

### 1.1. An example in 1D

To understand the difficulties of OSD problem let us consider the problem of design a string of thickness  $\alpha$  and length  $s$  for a musical instrument which gives a response as close to  $\psi_d$  as possible:

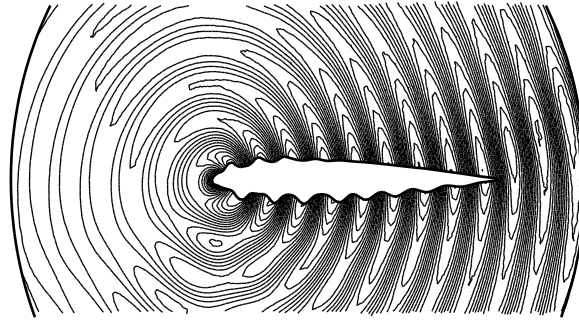


Fig. 2. Optimal design of an airfoil to minimize in a sector, with angle between  $180^\circ$  and  $225^\circ$ , the reflection of a monochromatic incident radar wave. The optimal shape is not admissible from the aerodynamic view point; a multi-disciplinary optimization is necessary (computed by A. Baron).

Find  $s > b$  and  $\alpha > 0$  such that

$$\min_{\alpha, s \in \mathcal{S}_d} \left\{ \int_a^b |\psi - \psi_d|^2 : \psi + \frac{d}{dx} \left( \alpha \frac{d\psi}{dx} \right) = f \text{ in } (0, s), \quad \psi(0) = 0, \quad \psi(s) = \psi_d(s) \right\}.$$

To discretize the problem, take  $I$  mesh points  $x_i$ , call  $\psi_i \equiv \psi(x_i)$ ,  $\delta x_{i+1/2} = x_{i+1} - x_i$ ,  $\delta x_i = (x_{i+1} - x_{i-1})/2$ ,  $x_I = s$  and consider

$$\min_{\delta x_i \in \mathcal{D}_{\mathcal{S}_d}} \left\{ \sum_{i \in [i_a, i_b]} |\psi_i - \psi_{d_i}|^2 \delta x_i : \psi_i + \frac{1}{\delta x_i} \left[ \alpha_{i+1/2} \frac{\psi_{i+1} - \psi_i}{\delta x_{i+1/2}} - \alpha_{i-1/2} \frac{\psi_i - \psi_{i-1}}{\delta x_{i-1/2}} \right] = f_i \right\}$$

with  $\psi_0 = 0$ ,  $\psi_I = \psi_{d_I}$ , i.e., with  $\Psi = \psi - \psi_d$ ,

$$\min_{\vec{\delta x} \in D} \{ \Psi^T B(\vec{\delta x}) \Psi : A(\vec{\alpha}, \vec{\delta x}) \Psi = F \}.$$

It is seen here that the problem is akin to control in the coefficient of PDEs [8], that the discrete problem has a greater unknown space than the continuous problem because the mesh comes also as a degree of freedom.

Nevertheless the problem is differentiable and so gradient methods should work. For this we will need the derivatives of the cost function with respect to all the unknowns,  $s, \alpha, \delta x_i$ . This can be a momentous task and so whenever possible automatic differentiation is of great help (see Refs. [6,5], and the bibliography therein).

## 2. Principle of automatic differentiation

Consider the problem of finding  $J'(u)$  when  $j(u)$  is given by a computer program. Because the program is made of differentiable lines,  $J'$  can be computed by differentiating every line and adding them to the computer program immediately above each line. For instance

```

Program for  $J$    lines to add to the program
 $x = 2u(u + 1)$     $dx = 2 * u * du + 2 * du * (u + 1)$ 
 $y = x + \sin(u)$   $dy = dx + \cos(u) * du$ 
 $J = x * y$         $dJ = dx * y + x * dy$ 

```

If this new program is run with  $u=u_0$ ,  $du=1$ ,  $dx=0$ ,  $dy=0$ ,  $dJ=0$ , then  $dJ$  is the derivative of  $J$  with respect to  $u$  at  $u_0$ .

### 2.1. Automatic AD

However, differentiating each line can be long and tedious. It can be done by the compiler by overloading the operators of arithmetic and the functions in the standard C-library. Operator overloading is available in Fortran 90 and C++. We present briefly an implementation in C++ because it seems difficult to make an efficient implementation in Fortran 90 [16]. Consider the following procedures:

- Step 1: if the program is in FORTRAN use f2C from <http://www.netlib.org/f2c/>
- Step 2: change float and double into ddouble and add #include ddouble.h

### 2.2. The library ddouble

Each variable has now two fields: its value and the value of its derivative. So we define a class of differentiable variables and stores these values in  $v$ . Every time an arithmetic operation is done, the corresponding operation on the derivatives must be done too. For instance below we give the overloading of the multiplication and of the addition:

```

class ddouble{ public:
    float v[2];
    ddouble(double a, double b = 0)
        { v[0] = a; v[1] = b; }
    // initialise la derivee a 0 sauf si b != 0
    friend ddouble operator *
        (const ddouble& a, const ddouble& b)
        { ddouble c;
          c.v[1] = a.v[1] * b.v[0] + a.v[0] * b.v[1];
          // (fg)' = f'g + fg'
          c.v[0] = a.v[0] * b.v[0];
          return c;
        }
}

```

```
friend ddouble operator+
  (const ddouble& a, const ddouble& b)
  { ddouble c;
    c.v[1] = a.v[1] + b.v[1]; // (f + g)' = f' + g'
    c.v[0] = a.v[0] + b.v[0];
    return c;
  }
// ...
} ;
```

### 3. Well posedness

Consider the academic problem of designing a wind tunnel with required flow properties in a region of space  $D$  (see Fig. 3). With a stream function formulation this would be

$$\min_{S \in \mathcal{S}_d} \left\{ \int_D |\psi - \psi_d|^2: -\Delta\psi = 0, \quad \psi|_S = 0 \quad \psi|_{\partial C} = \psi_d \right\}. \tag{1}$$

Before attempting any numerical simulation we may study the existence of solutions. These are by no means nonpractical questions because many of these optimal shape design problems do not have solutions. For example the optimization of a hook, clamped to the wall on the left and pulled so by weight on the right. With respect to weight under a given max constraints so that the structure does not break: optimal structures are of composite materials; there is no simply connected solution to this problem.

Although existence can be studied directly by using continuity results with respect to domain boundaries, one may also map the unknown domain from a fixed domain and consider that the unknown is now  $T: C \rightarrow \Omega$ ,

$$\min_{T \in \mathcal{T}_d} \left\{ \int_D |\psi - \psi_d|^2: \nabla \cdot [A \nabla \psi] = 0 \text{ in } C \quad \psi|_{\partial C} = \psi_d, \quad A = T'^{-1T} T'^{-1} \det T' \right\} \tag{2}$$

or extend the operators by zero in  $S$  and take the characteristic function for unknown:

$$\min_{\chi \in \mathcal{X}_d} \left\{ \int_D |\psi - \psi_d|^2: -\nabla \cdot [\chi \nabla \psi] = 0, \quad \psi(1 - \chi) = 0 \quad \psi|_{\partial C} = \psi_d \right\}. \tag{3}$$

This last approach, suggested in Ref. [15] has led to *topological optimization*.

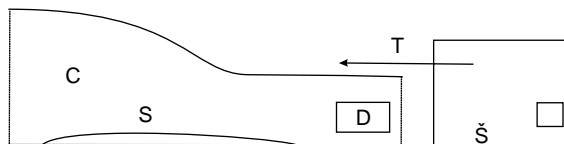


Fig. 3. Inverse design for a wind tunnel with desired properties  $\psi_d$  in  $D$ .

### 3.1. Results

Most results are obtained by considering minimizing sequence  $S^n$  and (in the case of our academic example) show that  $\psi^n \rightarrow \psi$  for some  $\psi$ , which is the solution of the PDE.

By using directly regularity with respect to the domain, Chenais [3] (see also Ref. [10]) showed that in the class of all  $S$  uniformly Lipschitz, problem (1) has a solution.

Similarly Murat-Simon [9] working with (2) showed that in the class of  $T \in W^{1,\infty}$  uniformly, the solution exists.

However, working with (3) generally leads to weaker results because if  $\chi^n \rightarrow \chi$ ,  $\chi$  may not be a characteristic function; one is lead to a *relaxed problem*.

In two-dimensional and for the Dirichlet problem there is a very elegant result due to Sverak [14]: *assume all admissible domains  $\Omega$  are included in a compact set  $D$  and the number connected components of  $D \setminus \Omega$  is less than a given number, then, in this class of sets, there exists a solution.*

### 3.2. Well-posedness by regularization

Another way to insure well posedness is to *regularize* the problem by changing the criteria and adding a “cost” to the control. For problem (1)

$$J(\Omega) = \int_D (\psi - \psi_d)^2 + \varepsilon \int_S dx$$

insures existence. More generally one may consider working with

$$J(\Omega) = \int_D (\psi - \psi_d)^2 + \varepsilon \|S\|_?^2$$

but the choice of norm is a delicate one. In general for second-order problem anything related to the second derivatives would be likely to work, but it is not known if weaker norms would work too.

## 4. Sensitivity analysis

Even though AD can solve the problem it is wise to check differentiability analytically. This can be done by using normal variation on a reference shape (see Fig. 4)

$$\partial\Omega^\alpha = \{x + \alpha \mathbf{n} : x \in \partial\Omega\}. \tag{4}$$

Following Cea [2], and Delfour-Zolezio [4] one may consider a velocity of deformation  $V(x)$  and define a time dependent shape

$$\Omega(t) = \{x + V(x)t : x \in \Omega\}$$

and compute  $dJ/dt$ , known as the *material derivative* of  $J$ .

Lately, for Neumann problems, the concept of topological derivative was introduced in Ref. [13]. One digs a small circular hole of center  $x = 0$  in the domain and study the limit of

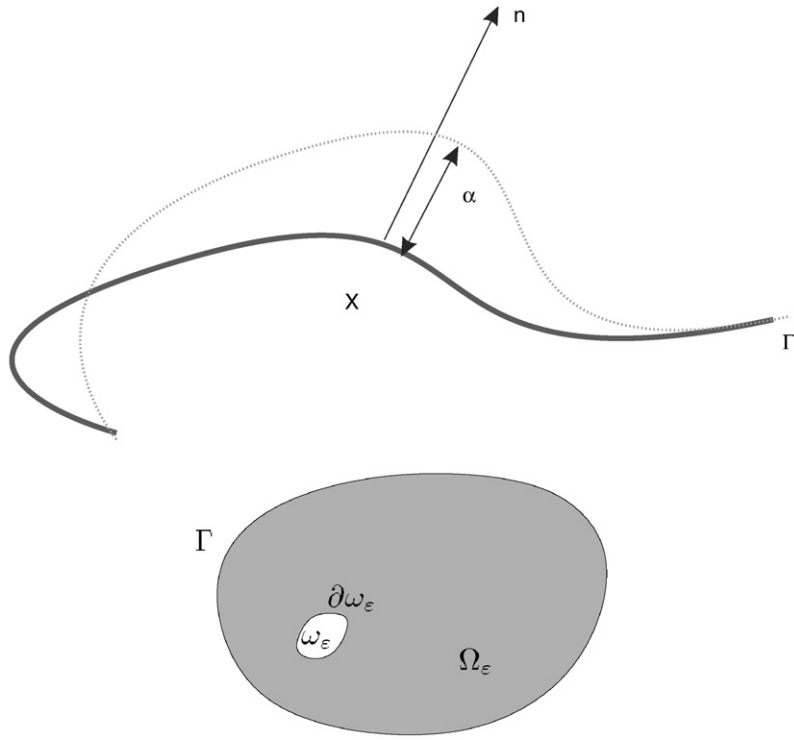


Fig. 4. Normal variations on a reference shape (top). Topological variation on the shape (bottom).

$(1/\varepsilon)(\psi^\varepsilon - \psi)$  where  $\psi^\varepsilon$  is the solution of the PDE with the hole and  $\psi$  the solution without the hole (see Fig. 4).

#### 4.1. Sensitivity: example

For the Laplace equation with Dirichlet conditions

$$-\Delta\psi^{\varepsilon\alpha} = f \text{ in } \Omega^{\varepsilon\alpha} \quad \psi^{\varepsilon\alpha} = 0 \text{ on } \Gamma^{\varepsilon\alpha} = \{x + \varepsilon\alpha n : x \in \Gamma\},$$

where  $\Omega^{\varepsilon\alpha}$  is obtained by (4) the derivative with respect to  $\alpha$  is calculated by assuming

$$\psi^{\varepsilon\alpha} = \psi + \varepsilon\psi'_\alpha + \frac{\varepsilon^2}{2}\psi''_\alpha.$$

By linearity  $\psi'$  and  $\psi''$  satisfy the PDE with zero rhs. By Taylor expansion:

$$0 = \psi^{\varepsilon\alpha}(x + \varepsilon\alpha n) = \psi^{\varepsilon\alpha}(x) + \varepsilon\alpha \frac{\partial\psi^{\varepsilon\alpha}}{\partial n}(x) + \frac{\varepsilon^2\alpha^2}{2} \frac{\partial^2\psi^{\varepsilon\alpha}}{\partial n^2}(x) + \dots$$

Therefore

$$-\Delta\psi'_\alpha = -\Delta\psi''_\alpha = 0 \quad \psi'_\alpha|_\Gamma = -\alpha \frac{\partial\psi}{\partial n} \quad \psi''_\alpha|_\Gamma = -\alpha \frac{\partial\psi'_\alpha}{\partial n} - \frac{\alpha^2}{2} \frac{\partial^2\psi}{\partial n^2}.$$

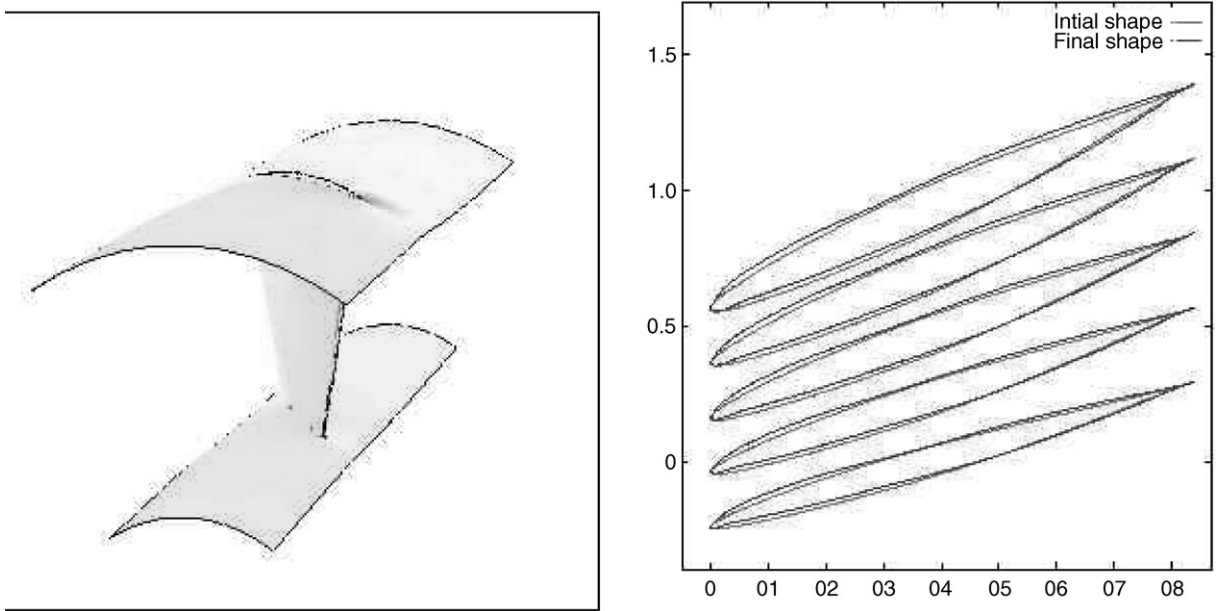


Fig. 5. Optimization of a cooling fan for a car engine. This three-dimensional optimization improved the design by 10%. The picture displays the final shape and the change at some cross section from the original hand optimized original design.

## 4.2. Navier–Stokes equations

### 4.2.1. The minimum drag problem

$$E(\Omega) \equiv \min_{\Omega \in C} \int_{\Omega} \frac{1}{2} \|\nabla u\|^2 dx: \quad u|_{\partial\Omega} = g$$

$$u \nabla u + \nabla p - \nu \Delta u = 0, \quad \nabla \cdot u = 0,$$

Sensitivity analysis by local variations gives

$$\partial\Omega' = \{x + \alpha n(x) \in \partial\Omega\} \Rightarrow \delta E = \int_{\partial\Omega} \chi \alpha ds + o(|\alpha|).$$

$$\delta E = \int_{\Omega} \nabla u \cdot \nabla \delta u + \frac{1}{2} \int_{\partial\Omega} \alpha |\nabla u|^2 = \frac{1}{2} \int_{\partial\Omega} \alpha \partial_n u \cdot (\partial_n u + 2 \partial_n w) + o(|\alpha|)$$

$$-\mathbf{u} \nabla \mathbf{w} + \mathbf{w} \nabla \mathbf{u}^T + \nabla \mathbf{q} - \nu \Delta \mathbf{w} = \mathbf{u} \nabla u, \quad \nabla \cdot \mathbf{w} = 0, \quad \mathbf{w}|_{\partial\Omega} = \mathbf{0}.$$

More can be found in Refs. [11,12]. See also Fig. 5.



### 4.3. Gradient methods

So one starts with a smooth shape, moves each point in its normal direction by

$$\alpha = \partial_n u \cdot (\partial_n u + 2\partial_n w).$$

But will the new shape have the same regularity? In general the answer is no, and this loss of regularity is numerically dangerous and prone to generation of oscillations. The cure is to use a *smother* which mathematically amounts to change the norm of the gradient method. For instance if the shape is moved by  $\beta$  solution of

$$-\frac{d^2\beta}{ds^2} = -\alpha$$

implies then

$$\begin{aligned} j(S(\beta)) - j(S) &= \frac{1}{2} \int_{\partial\Omega} \beta \partial_u \cdot (\partial_n u + 2\partial_n w) \\ &= -\frac{1}{2} \int_{\partial\Omega} \beta \alpha = - \int_{\partial\Omega} \left| \frac{d\alpha}{ds} \right|^2. \end{aligned}$$

### 4.4. Discretization

Consider again the academic problem (1). It can be discretized by

$$\min_{q_h} \left\{ j_h(q_h) = \int_D |\psi - \psi_d|^2 : \int_{\Omega} \nabla \psi_h \nabla w_h + \frac{1}{\varepsilon} \int_{\Gamma} (\psi_h - \psi_d) w_h = 0 \quad \forall w_h \in V_h \right\},$$

where  $V_h$  is the finite element space of piecewise linear continuous functions.

Calculus of variation is possible but the degree of freedoms are now the node motion  $q_h \in V_h$ , a piecewise linear continuous function built from its values at the vertices, namely the motions of the same (see Fig. 6). Let

$$\int_{\Omega} \nabla p_h \nabla w_h + \frac{1}{\varepsilon} \int_{\Gamma} p_h w_h = 2 \int_D (\psi_h - \psi_d) w_h \quad \forall w_h \in V_h.$$

Finally

$$\delta J_h = \int_{\Omega} \nabla \psi_h (\nabla \delta q_h + \nabla \delta q_h - \nabla \cdot \delta q_h)^T \nabla p_h \approx \int_{\Gamma} \nabla \psi_h \nabla p_h \delta q_h \cdot n.$$

Recognize here the linearization of  $\nabla Q_h \nabla Q_h^T \det \nabla Q_h$  at  $x + q_h(x)$ , therefore

**Corollary 1.** *The change of cost function due to inner nodes is  $O(h^2)$*

### 4.5. Guidelines

To our experience success in solving an OSD problem depends on the following.

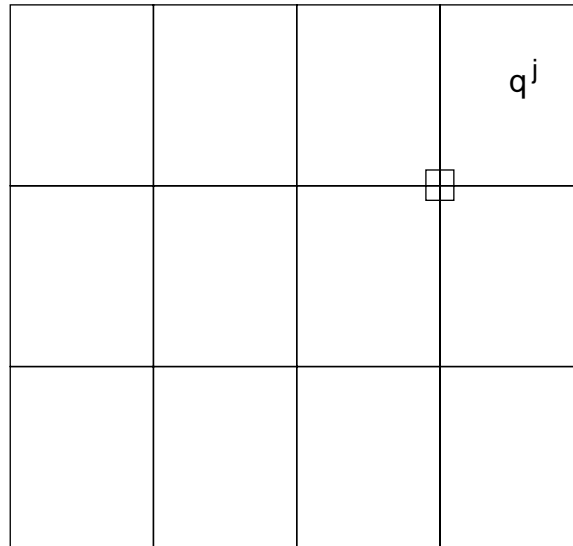


Fig. 6. After discretization, not only the boundary vertices but also the inner vertices are the degrees of freedom of the optimization problem.

- Whenever possible second-order optimization methods (BFGS for instance) should be used because the problems are stiff.
- Compute derivatives with respect to boundary nodes only and apply the theory of approximate gradients to combine mesh refinement with optimization (see Ref. [7] for example).
- Use a smoother, i.e. do not work with the  $L^2(\Gamma)$  norm of the node displacements. Use it also to move the inner vertices.
- Experience also shows that

$$\delta_S j(\Omega) = \delta_S \int_S F(\psi) \cdot n \approx \int_S F(\psi) \cdot \delta n + \dots .$$

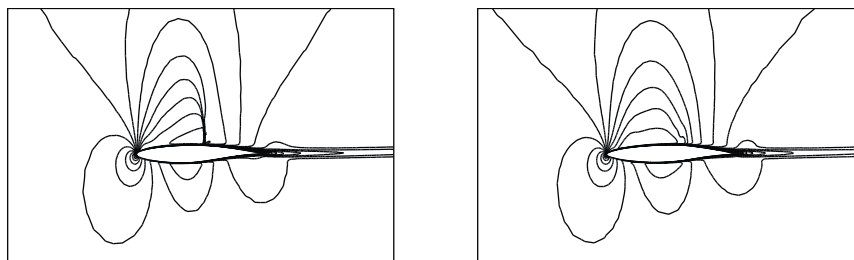
Namely the most important variation is due to the change of the normal in the case of surface integrals.

## 5. Implementation issues and results

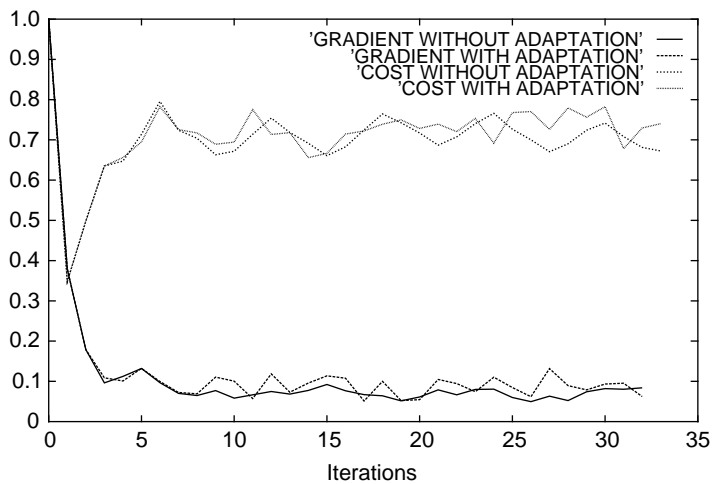
### 5.1. Link with CAD

In industries, shapes are stored in Computed Aided Design data bases as a set of Bezier patches or others with infinite details such as screws and bolts which are not relevant to a finite element calculation in aerodynamics for instance. Furthermore the CAD system is proprietary.

Therefore it is convenient to abstract the optimization from the CAD system and ask the engineer for any triangulation of the surface and use it as our initial design. The strategy is then what we



Normalized convergence histories for the cost and gradient with and without adaptation



Initial and optimized shapes with and without adaptation

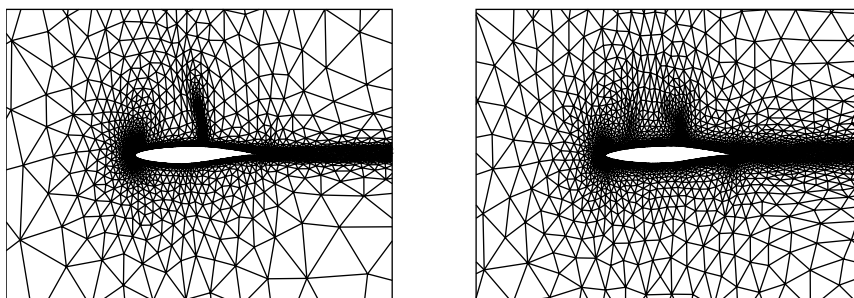
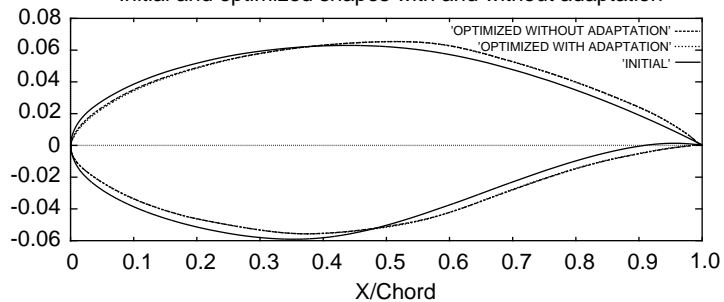


Fig. 7. Optimization of a wing profile.

call a *CAD-free optimization platform*:

- (1) Generate any surface mesh from the CAD data.
- (2) Apply a  $C^1+$  edges recognition software for surface mesh refinement.
- (3) Apply a three-dimensional volumic automatic mesh generator from the surface mesh.
- (4) Do the optimization with mesh refinement using the same module as in step 2 but coupled with the PDE solver.
- (5) Feed back the result into the CAD system.

### 5.2. Optimization of a wing profile

See Fig. 7.

Drag is mostly pressure drag due to the shock (pressure drag). The lift and area are imposed by a penalty method with parameters  $\varepsilon, \beta$ .

$$J(u, p, \theta) = F \cdot u_\infty + \frac{1}{\varepsilon} |F \times u_\infty - C_l|^2 + \frac{1}{\beta} \left( \int_S dx - a \right)^2$$

with  $F = \int_S (p \mathbf{n} + (\mu \nabla u + \nabla u^T))$  and a Navier–Stokes +  $k - \varepsilon$  + wall laws flow solver.

## 6. Prospective

Recently the optimization of a complete aircraft (a business jet) was done with this method (even on a workstation when incomplete gradients are used) and a 10% improvement obtained after a few iterations.

However OSD is still a difficult and computer intensive task. There is a good prospect for global nondifferentiable optimization because there are often many local minima and because the flow solver is often available in binary format only (such would be the case if a commercial software was used). However, genetic algorithms are still slow and difficult to couple with gradient methods.

Incomplete gradients is also a good field of research and awaits mathematical proofs.

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### **Further reading**

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