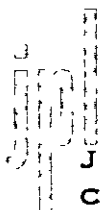
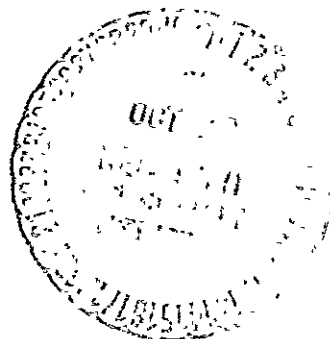


N66-23501

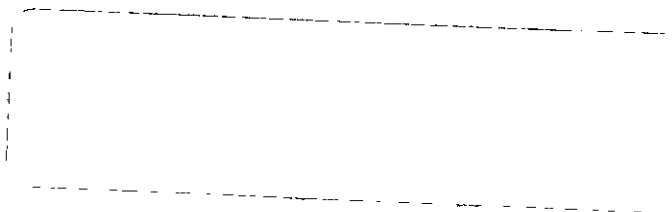
Technical Memorandum No. 33-204
***SPODP - Single Precision Orbit
Determination Program***

Michael R. Warner
Melba W. Nead



**JET PROPULSION LABORATORY
CALIFORNIA INSTITUTE OF TECHNOLOGY
PASADENA, CALIFORNIA**

February 15, 1965



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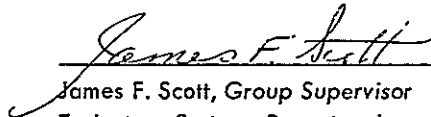
50759302K 26

Technical Memorandum No. 33-204

*SPODP – Single Precision Orbit
Determination Program*

Michael R. Warner

Melba W. Nead


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Trajectory Systems Programming


H. Fred Lesh, Manager
Computer Applications

**JET PROPULSION LABORATORY
CALIFORNIA INSTITUTE OF TECHNOLOGY
PASADENA, CALIFORNIA**

February 15, 1965

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National Aeronautics & Space Administration

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ABSTRACT

This Technical Memorandum describes the Single Precision Orbit Determination Program which was developed for operation under the data processing system of the Space Flight Operations Facility. Included are sections containing flow diagrams, load maps, the common storage map, input and output descriptions, subroutine writeups, operating instructions, and check cases. The computational methods employed by the program are described in the subroutine documentation.

I. INTRODUCTION

The Single Precision Orbit Determination Program (SPODP; more commonly expressed as ODP) for the IBM 7094 computer was written to meet the specifications of the Jet Propulsion Laboratory for a reliable and accurate method of tracking and predicting the motion of lunar and interplanetary spacecraft. The uses of the ODP may conveniently be separated into real-time and nonreal-time applications.

Real-time applications:

1. To establish a reliable set of orbital elements for the spacecraft.
2. To provide an acquisition ephemeris for the world-wide network of tracking stations.
3. To assist JPL engineers in evaluating the performance of tracking stations and the quality of tracking data.

Nonreal-time applications:

1. To provide a high-speed computing method necessary for orbit determination and tracking data accuracy studies (pre-mission).
2. To provide a high-speed computing method necessary for a sophisticated orbital analysis based on large numbers of observations (post-mission).

In addition to solving for the six initial conditions of the spacecraft's motion, the ODP has the capability of solving for 14 physical constants and the Earth radius, latitude, and longitude of 15 tracking stations. From this set of 66 parameters, a subset containing from one to twenty is extracted by the user. The ODP obtains solutions for the parameters in this subset.

The ODP, since it must have initial estimates of the parameters, is an orbit improvement program. It differentially connects the estimates by means of an over-determined system of equations, employing a modified least-squares method. This procedure may be briefly outlined as follows:

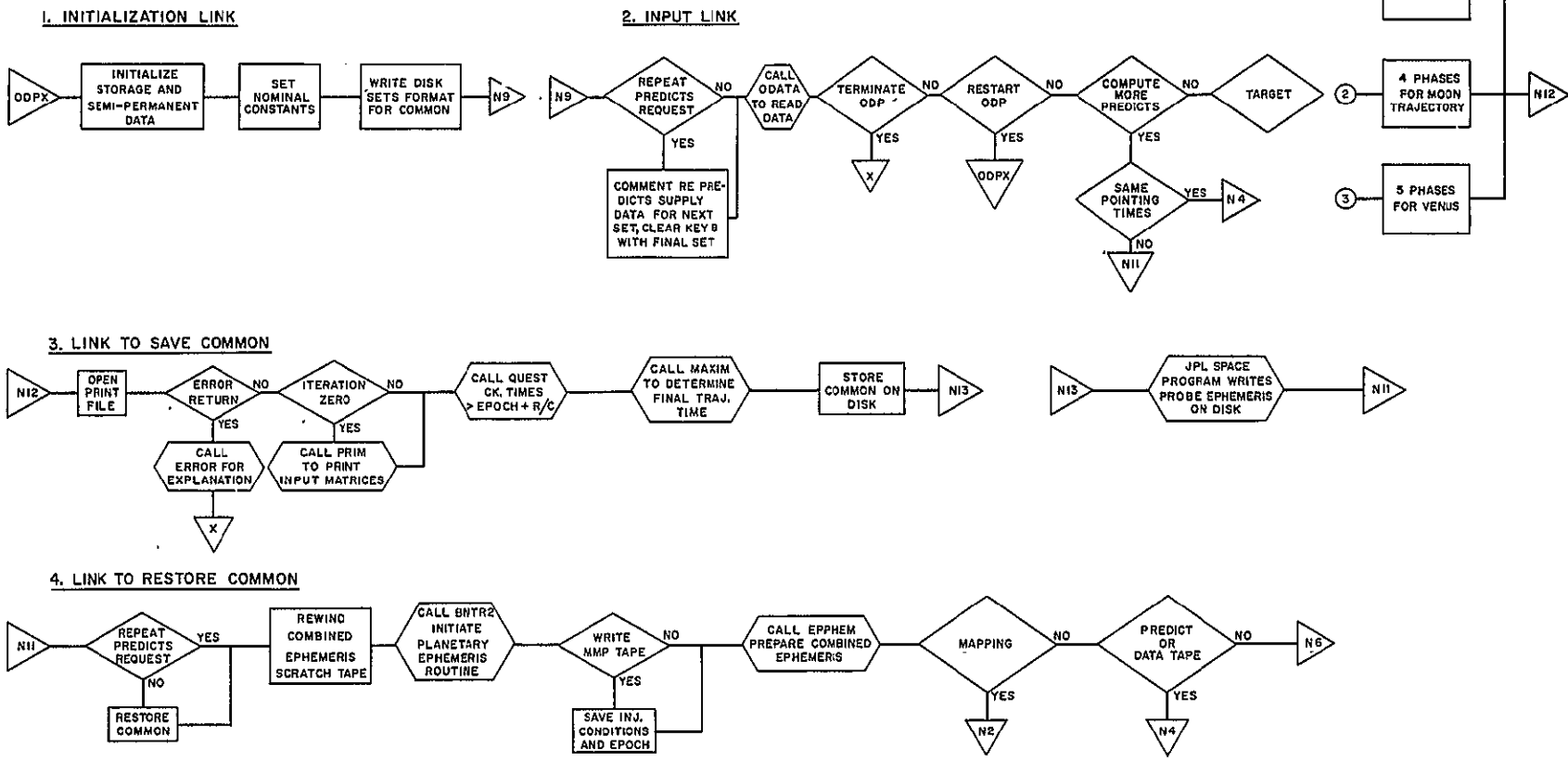
1. Input parameter estimates.
2. Write probe ephemeris file on disk based on orbit estimate.
3. Read i^{th} observation from data file, G_i . This observation may be slant range, range rate, one of four angle types, one of four doppler types, or DSIF ranging.
4. Using probe ephemeris, determine the value of the observation based on orbit estimate, F_i .
5. Obtain the residual $\Delta F_i = G_i - F_i$.
6. Calculate the partials of the observations with respect to the n parameters to be estimated, $\partial F_i / \partial Q_1, \dots, \partial F_i / \partial Q_n$.
7. Multiply the column matrix of partials by itself to form a matrix J_i^* .
8. Add J_i^* to the accumulated matrix $J^* = J_1^* + J_2^* + \dots, + J_{i-1}^*$.
9. Multiply the column matrix of partials by the residual to form a column matrix R_i .
10. Add R_i to the accumulated column matrix $R = R_1 + R_2 + \dots, + R_{i-1}$.
11. Repeat steps 3 through 10 until all observations are processed.
12. Obtain the final J matrix by adding in the a priori uncertainties. $J = J^* + \tilde{\Gamma}^{-1}$.
13. Solve the normal equations $J\Delta Q = R$ for the column matrix of changes to the parameter estimates, ΔQ .
14. Repeat steps 2 through 13 until the procedure converges.

The ODP is written on disk in thirteen links. Each link constitutes a logical section of the overall orbit determination. The linking is accomplished by use of the JPTRAJ source program, which is also used to pass certain information between links.

II. FLOW CHARTS

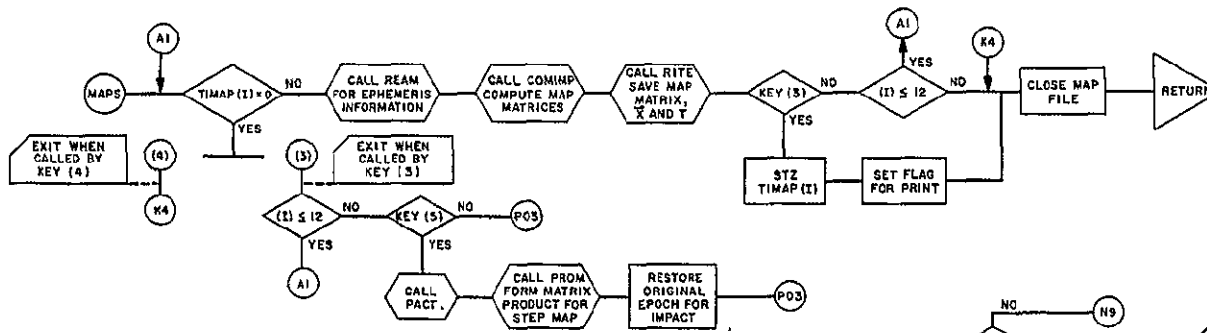
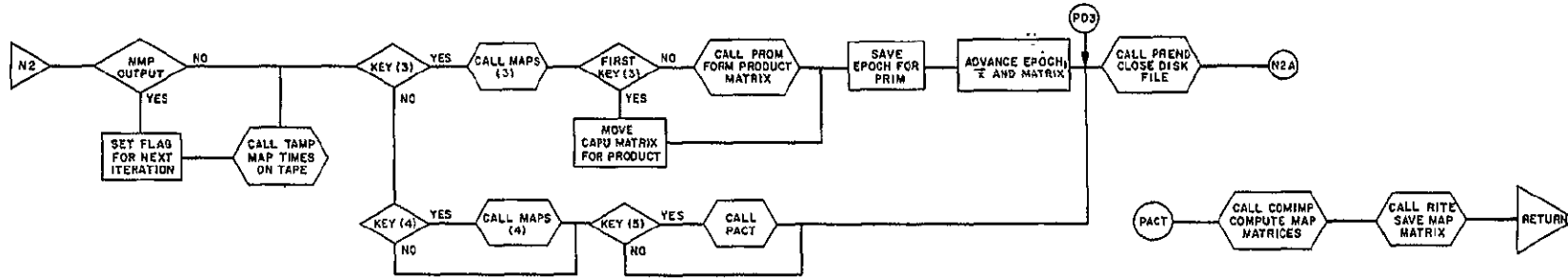
A. LINKS

1. Initialization Link
2. Input Link
3. Link to Save COMMON
4. Link to Restore COMMON
5. Link to Compute Mapping Matrices
6. Link to Output Mapping Matrices
7. Link to Compute Closest Approach Parameters
8. Predictions or Data Simulation Link
9. Link to Sort Predictions
10. Data Fitting Link
11. Link to Output Solution
12. Link to Output Residuals and Statistics
13. Link to Output Auxiliary Quantities

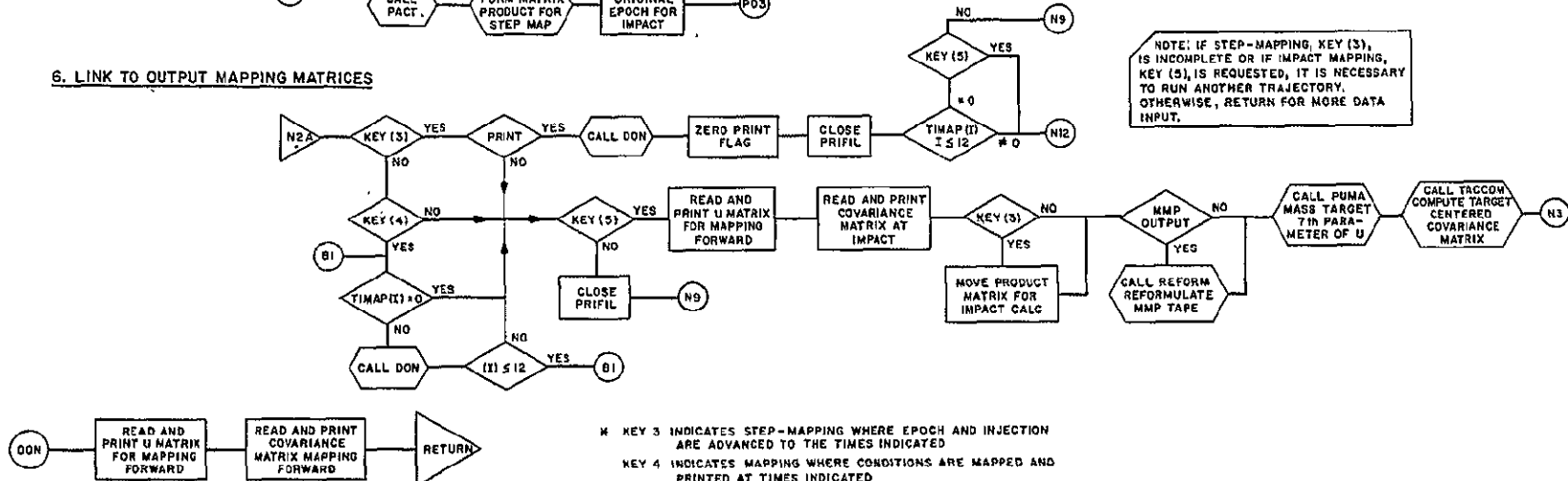


5

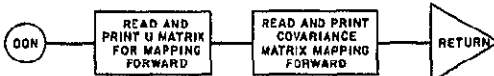
5. LINK TO COMPUTE MAPPING MATRICES



6. LINK TO OUTPUT MAPPING MATRICES

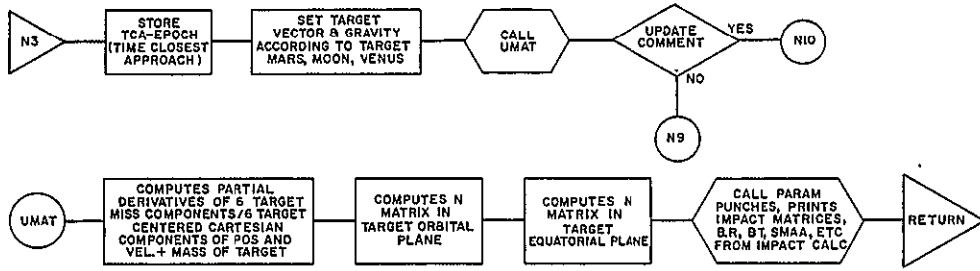


NOTE: IF STEP-MAPPING, KEY (3), IS INCOMPLETE OR IF IMPACT MAPPING, KEY (5), IS REQUESTED, IT IS NECESSARY TO RUN ANOTHER TRAJECTORY. OTHERWISE, RETURN FOR MORE DATA INPUT.

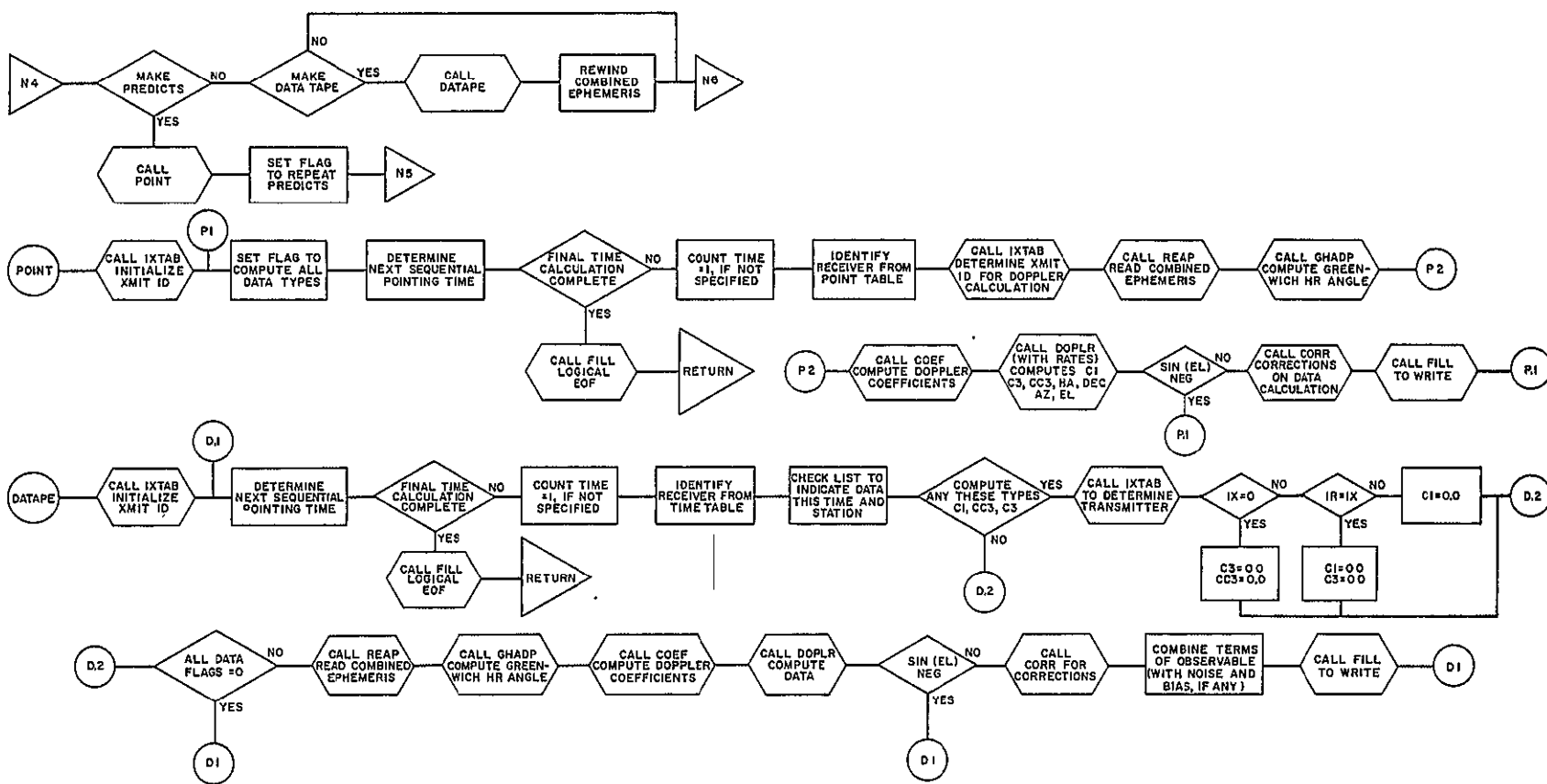


- * KEY 3 INDICATES STEP-MAPPING WHERE EPOCH AND INJECTION ARE ADVANCED TO THE TIMES INDICATED
- KEY 4 INDICATES MAPPING WHERE CONDITIONS ARE MAPPED AND PRINTED AT TIMES INDICATED
- KEY 5 INDICATES IMPACT MAPPING, CLOSEST APPROACH PARAMETERS ARE COMPUTED.

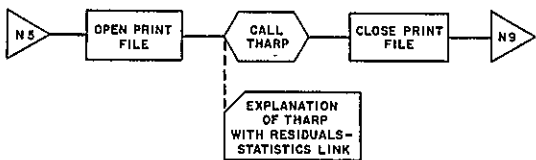
7. LINK TO COMPUTE CLOSEST APPROACH PARAMETERS



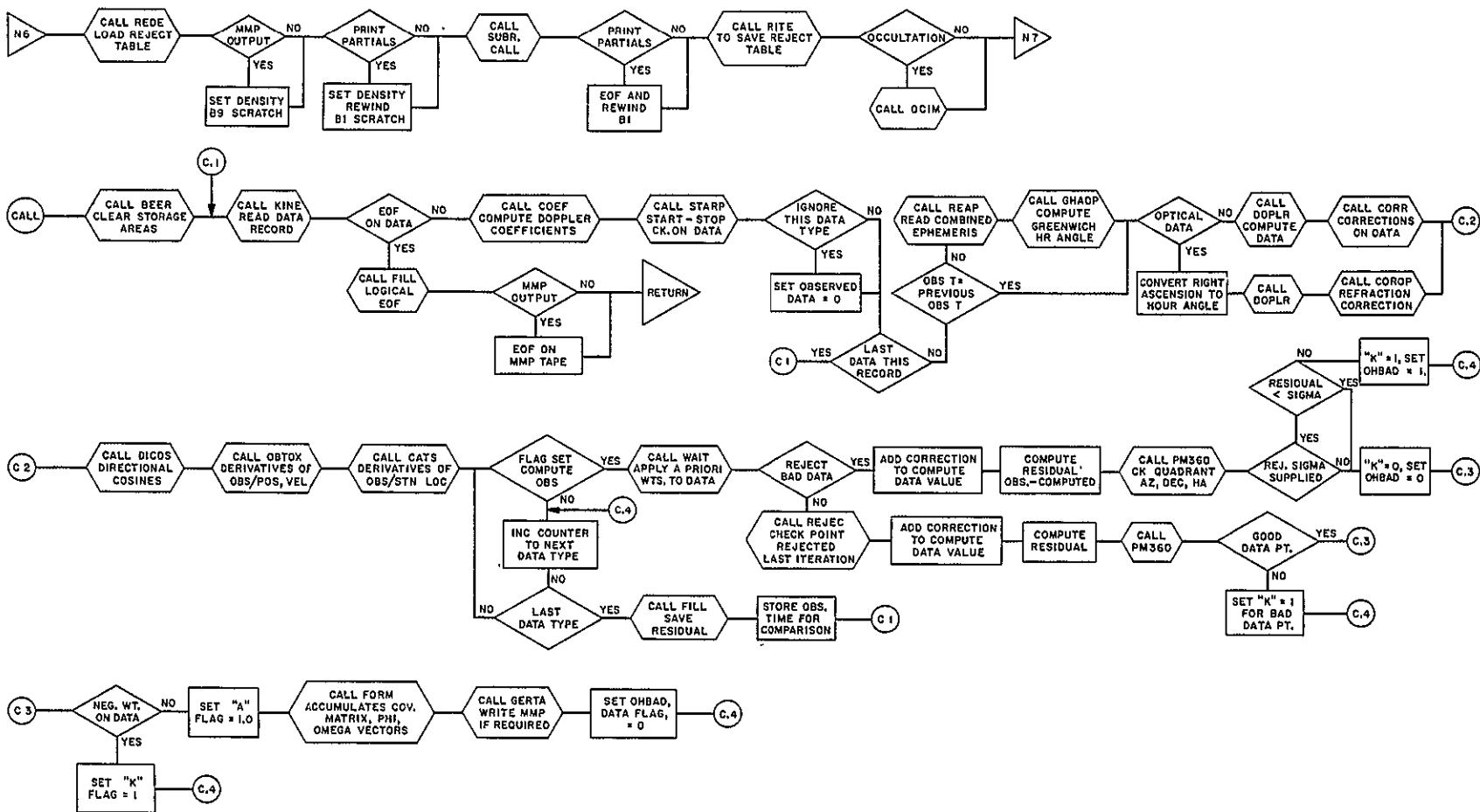
8. PREDICTIONS OR DATA SIMULATION LINK



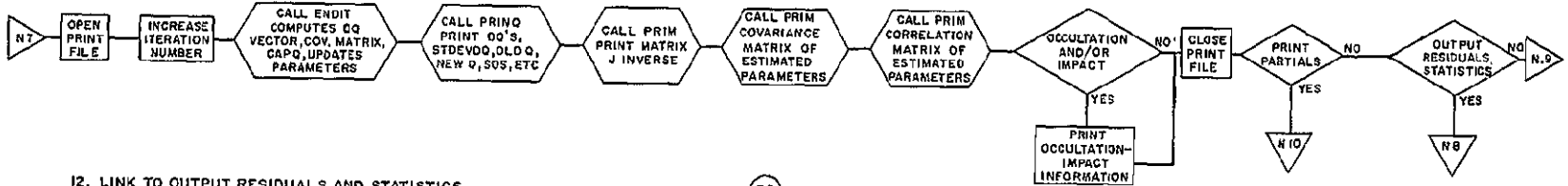
9. LINK TO SORT PREDICTIONS



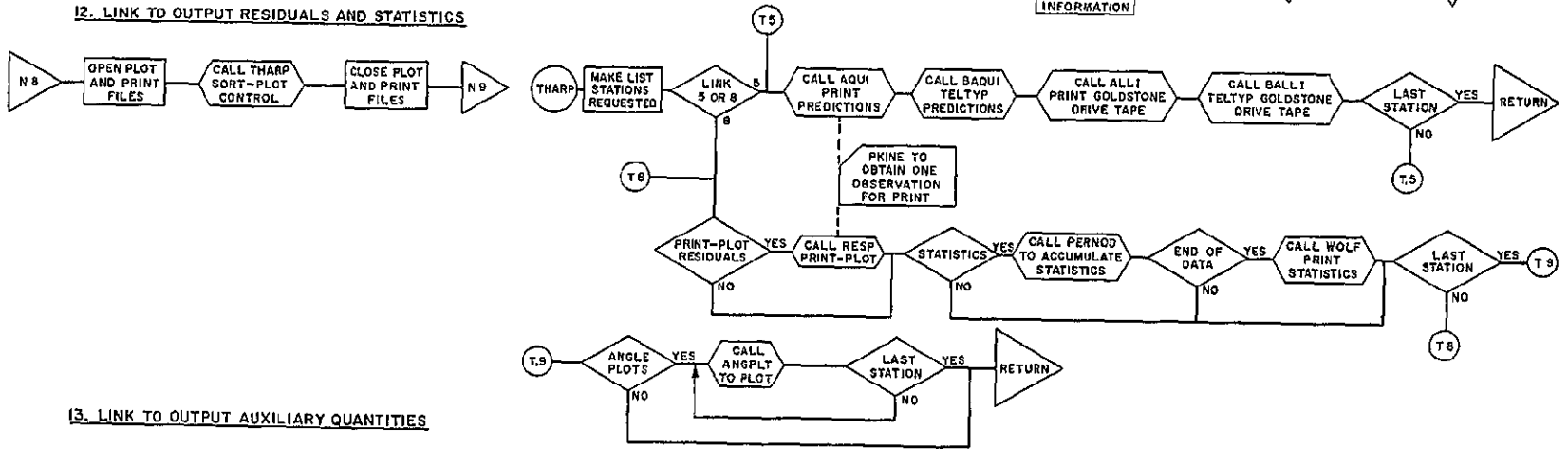
10. DATA FITTING LINK



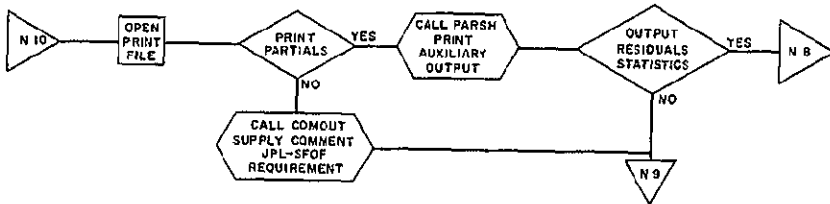
11. LINK TO OUTPUT SOLUTION



12. LINK TO OUTPUT RESIDUALS AND STATISTICS



13. LINK TO OUTPUT AUXILIARY QUANTITIES

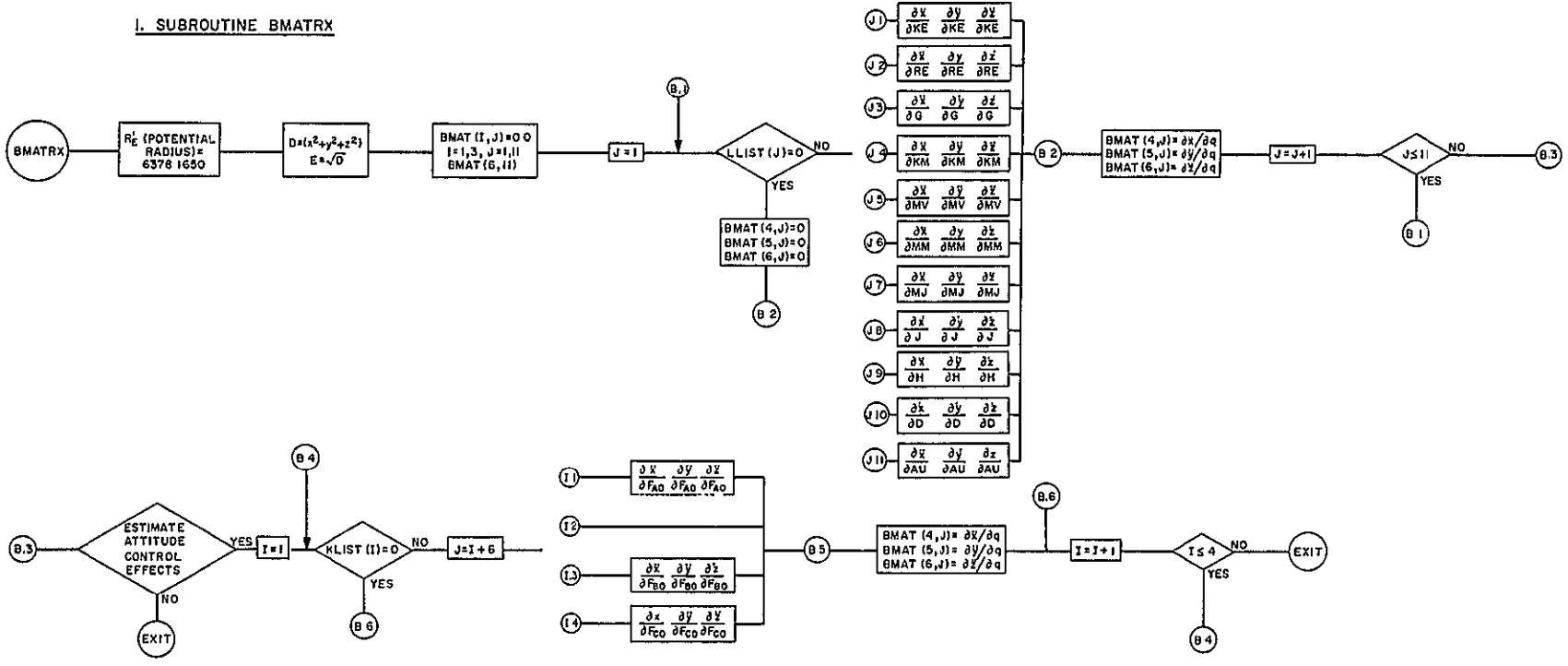


6

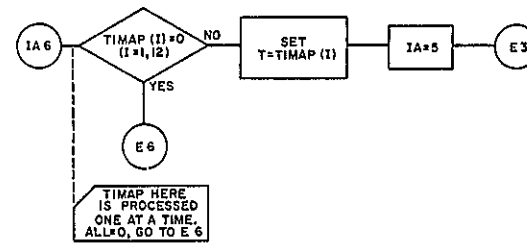
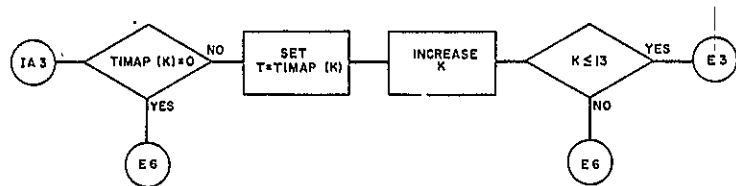
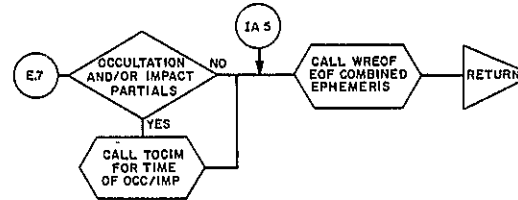
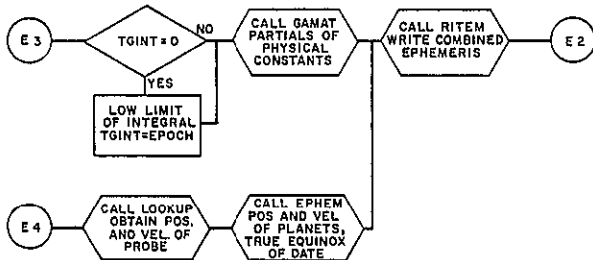
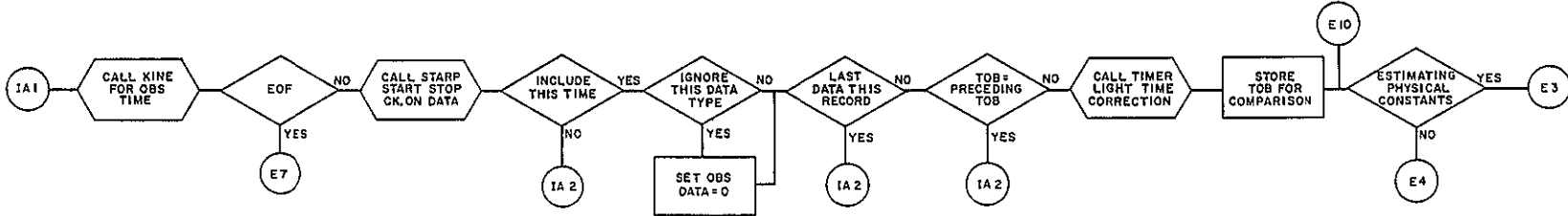
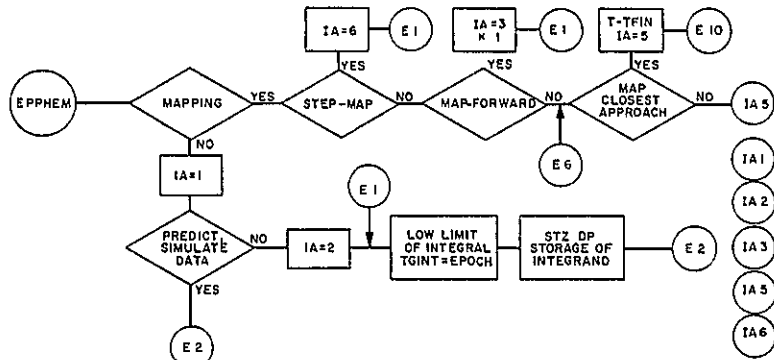
B. SUBROUTINES

1. BMATRX
2. EPPHEM
3. GAMAT
4. LOOKUP
5. MAXIM

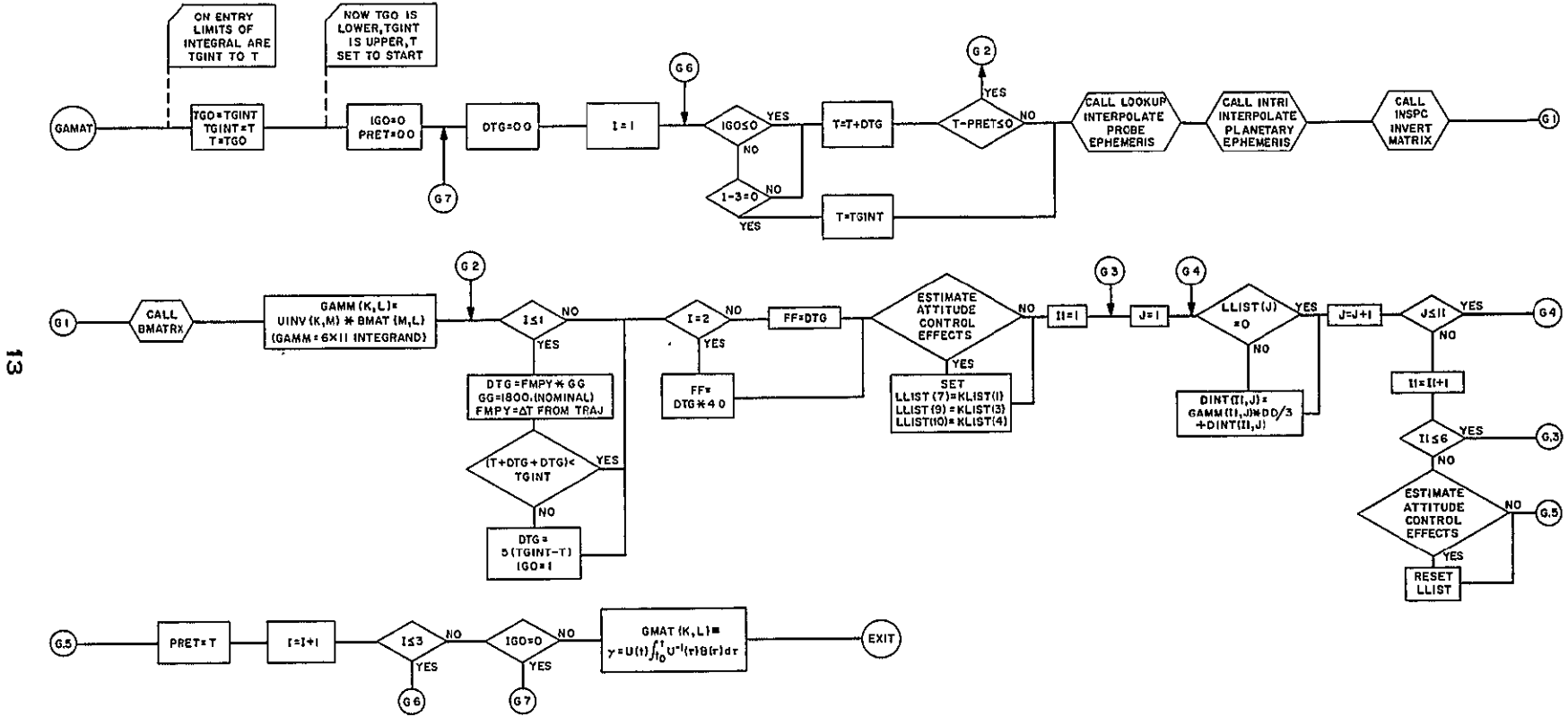
I. SUBROUTINE BMATRIX



2. SUBROUTINE EPPHEM

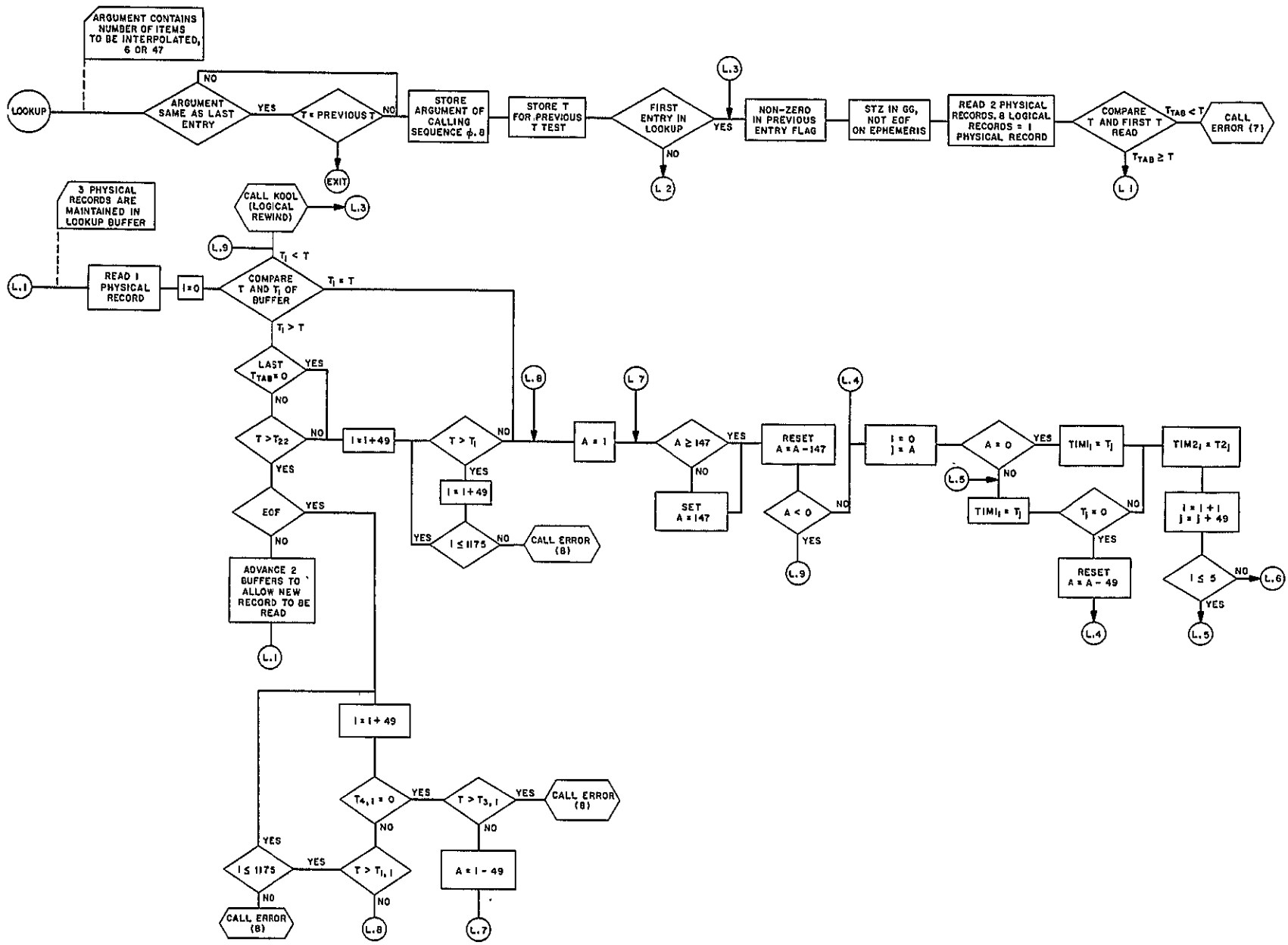


3. SUBROUTINE GAMAT



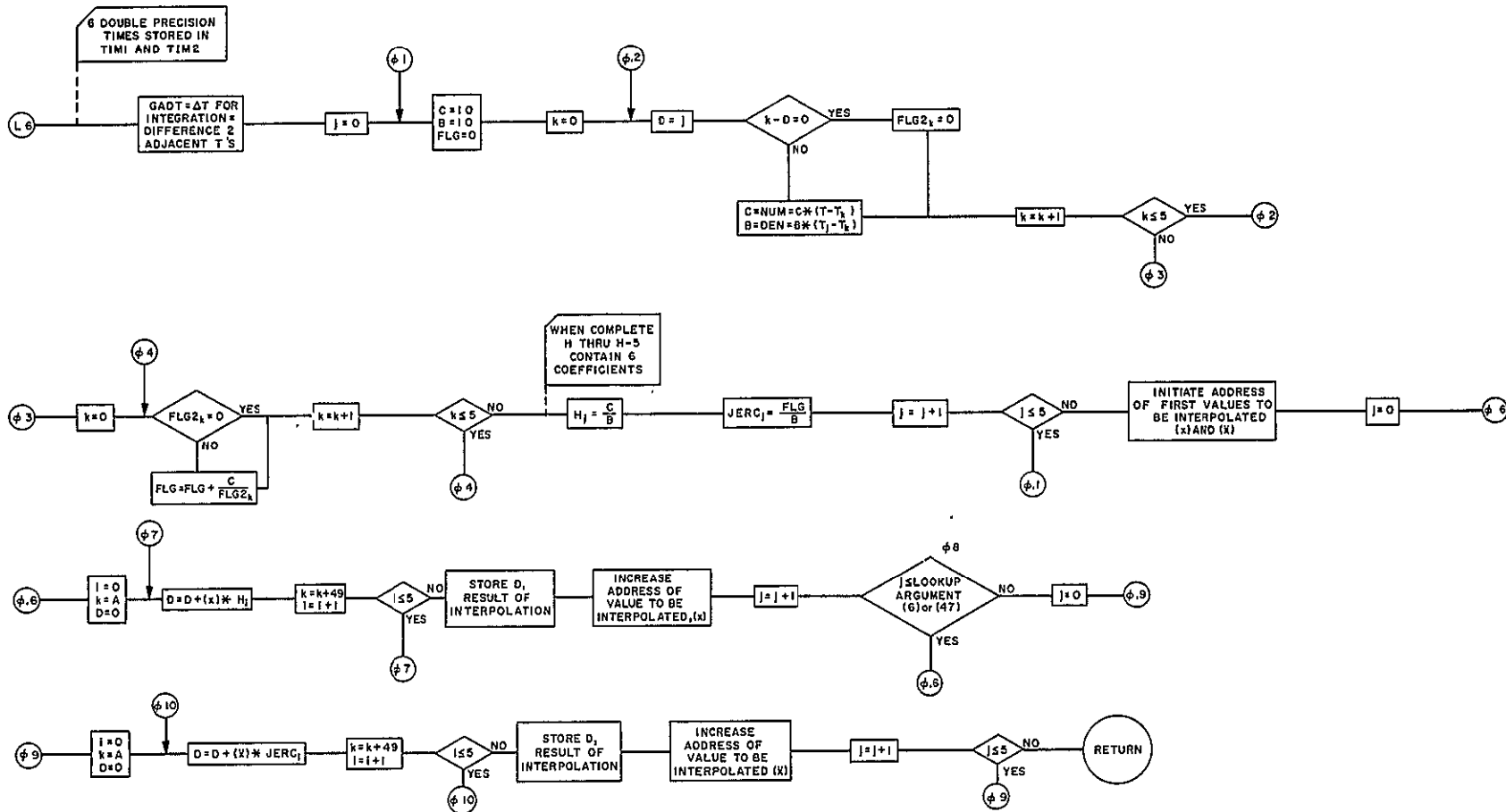
13

4. SUBROUTINE LOOKUP

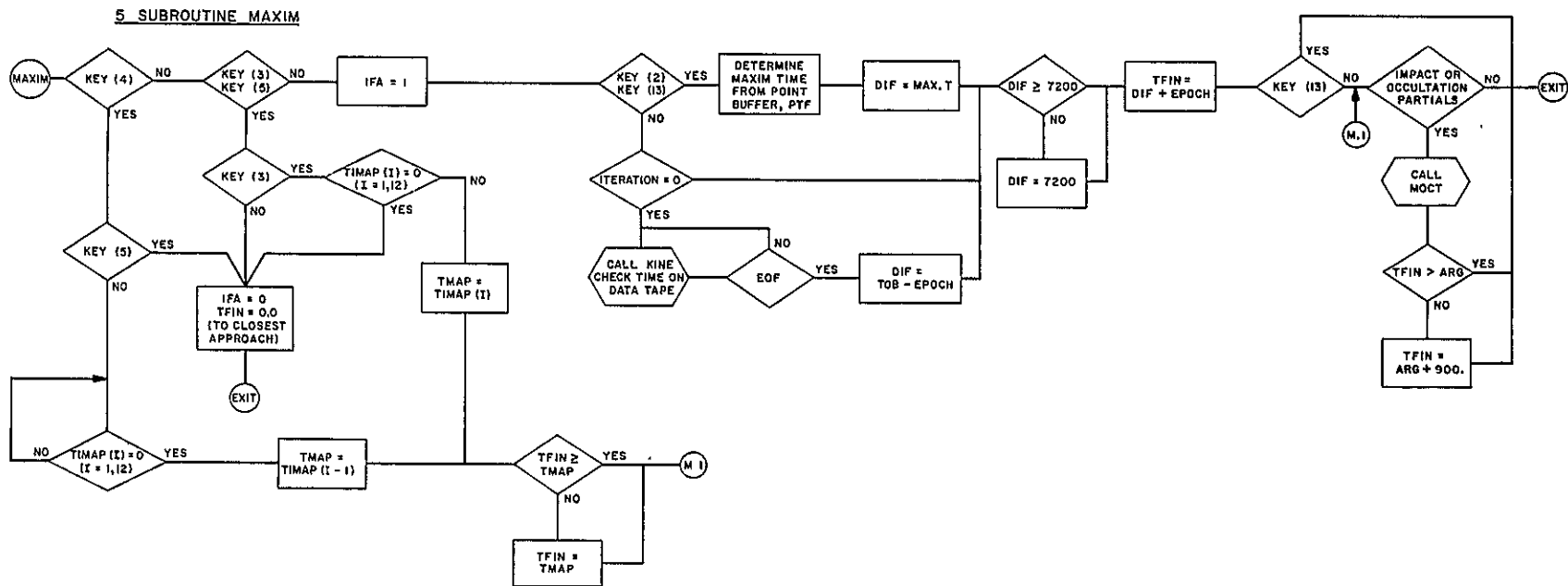


14

4. SUBROUTINE LOOKUP p 2



15



III. PHYSICAL LIMITATIONS

A. DISK AND TAPE USAGE UNDER SFOF

The ODP uses tape unit B6 for the planetary ephemeris (EPHEM) tape and tape unit A4 for the combined ephemeris scratch tape. Disk allocation is restricted to the 2800 record block named JPLODP.

B. DEBUGGING FACILITIES

Snaps may be inserted in links ODPX, LA2A, LA3, LA5, LA7, LA8, LA10, and LA12, as these links contain subroutine PROUT.

Octal correctors may be inserted in any link through the JPTRAJ source deck.

IV. INPUT AND STORAGE ALLOCATION

Primary input to the ODP is a deck of control and data cards, which are described in detail in Section IV. A. These cards contain the parameter estimates, the desired options, and all variables which are not a function of each observation. Subroutine CARDS is used for card input.

Also input to the ODP is a file on disk containing the tracking data, and a planetary ephemeris tape. The data file is written on disk by the Orbit Data Generator Program, ODG, while the ephemeris tape is the standard JPL EPHEM system tape. The system routine DCP is used for disk reading and subroutine TAPIO is used for reading the planetary ephemeris tape.

A. CARD INPUT

JPL TECHNICAL MEMORANDUM NO. 33-204

```

EPOCH
01
*
* EPOCH ASSOCIATED WITH THE TRAJECTORY, WHERE
M YY= LAST TWO DIGITS OF THE YEAR
* MM= MONTH OF THE YEAR
* 0= ZERO
* DD= DAYS
* HH= HOURS
* NN= MINUTES
* SS= SECONDS
* FFF= MILLISECONDS
YHM00DHH,N4SSFF

GEOCENTRIC POSITION AND VELOCITY AT EPOCH
102
*
* SYMBOLS ALLOWED
X,Y,Z,DX,DY,DZ

INPUT OTHER OPTIONS AND CONSTANTS
103
*
* SYMBOL ALLOWED EXPLANATION AND/OR NOMINAL
KE (EARTH) (398603.2KM3/SEC2)
RE EARTH RADIUS (6378.3149KM)
KM KINOMIN
MV MASS RATIO VENUS (2.447118E-6)
MH MASS RATIO MARS (3.2303420E-7)
MJ MASS RATIO JUPITER (9.5475420E-4)
J CJEF. SECOND HARMONIC (1.62345E-3)
H CJEF. THIRD HARMONIC (-.575E-5)
Q CJEF. FOURTH HARMONIC (1.7875E-5)
AJ ASTRONOMICAL UNIT (1.4959986KM)
C VELOCITY OF LIGHT (299792.5KM/SEC)
RI(1) STATION RANGE
LA(1) STATION LATITUDE (GEOCENTRIC)
LJ(1) STATION LONGITUDE
*
* RI(1)=6372.2599 LA(1)=-35.066620 LJ(1)=243.20507
*
* RI(2)=6372.0341 LA(2)=35.20807 LJ(2)=243.15082
*
* RI(3)=6371.6686 LA(3)=-35.219630 LJ(3)=148.98028
*
* RI(4)=6372.6040 LA(4)=-31.211875 LJ(4)=136.88727
*
* RI(5)=6375.4980 LA(5)=-25.739277 LJ(5)=27.685181
*
* RI(6)=6376.3091 LA(6)=17.0355 LJ(6)=298.2072
*
* RI(7)=6377.8013 LA(7)=-7.8991 LJ(7)=345.5876
*
* RI(8)=6376.1150 LA(8)=18.064939 LJ(8)=292.91122
*
* RI(9)=6375.6810 LA(9)=-25.7904 LJ(9)=28.3580
*
* RI(10)=6374.05 LA(10)=-24.75336 LJ(10)=113.71605
*
* RI(11)=6372.05 LA(11)=32.1709 LJ(11)=295.3465
*
* RI(12)=6371.6586 LA(12)=-35.21963 LJ(12)=148.98028
*
* RI(13)=6370.0868 LA(13)=40.23800 LJ(13)=355.7505
*
* RI(14)=6375.9450 LA(14)=-18.823066 LJ(14)=204.31472
*
* RI(15)=6180.1990 LA(15)=10.668 LJ(15)=0.0

NSDISP(1) NORTH-SOUTH DISPLACEMENT, DEG
EWDISP(1) EAST-WEST DISPLACEMENT, DEG
INDEX(1) INDEX OF REFRACTION (340.0)
DRIFT TRANSDUCER DRIFT, CPS/SEC
STEP INITIAL TRAJECTORY STEP SIZE, SEC
ENERGY POWERED FLIGHT ENERGY
DTBURV POWERED FLIGHT DURATION
MOTOR MOTOR COUNT
JETON JETON FOR A.C.
JETOFF JETOFF FOR A.C.
PREDID IDENTIFICATION FOR PREDICTS
READ ORBIT FROM DISK ORBIT STORED ON DISK IN LINK 6
READ MATRIX FROM DISK MATRIX STORED ON DISK IN LINK 6
SOLAR PRESSURE OFF
LIGHT TIME OFF
REFRACTION AND VERTICAL OFF
IMP OUTPUT
PHI VECTOR OUTPUT
UPDATE COMEVI TO WRITE DISK FOR M/C AND SPACE
DJT L.T.-U.T., SEC (35.)
FJELWT WEIGHT OF FUEL, LB.
RFUPL FLOW RATE, LB/SEC
THRUST THRUST, LB
BRNNG BIAS ANGLE, DEG
RSTOP TARGET RADIUS, KM (MARS,3400.0)
* (MOON,1738.09)
* (VENUS,6200.0)
RAND INPUT FOR RANDOM NUMBER, OCTAL (1)
RMAX MAXIMUM SLANT RANGE FOR DATA, DEG
ELMIN MINIMUM ELEVATION ANGLE FOR DATA, DEG
DATA SIGMA INPUT RESIDUAL PRINT REPLACED BY
* STANDARD DEVIATIONS
REM CONSTRAINT RE IS DETERMINED BY THE RELATION
* KEM= CODE H001 (86.315745(KC+KM))
* KC AND KM INITIALS ARE CONSISTENT
* WITH THE CONSTRAINT
RADOP? RADIATION PRESSURE OPTION (1.031E8)

FLRES NON-ZERO, RESIDUALS IN FLOATING PT.

PJMAT PUNCH MATRIX, N
1 INPUT J MATRIX OF ESTIMATED PARAMETERS
2 INPUT COVARIANCE MATRIX OF CONSIDERED PARAMETERS
3 J INVERSE
4 COVARIANCE MATRIX OF ESTIMATED PARAMETERS
5 CORRELATION MATRIX OF ESTIMATED PARAMETERS
6 COVARIANCE MATRIX AT IMPACT
7 (ALREADY PUNCHED)
8 J MATRIX
9 INPUT COVARIANCE MATRIX OF ESTIMATED PARAMETERS
10 CORRELATIONS BASED ON J MATRIX
11 U PRODUCT MATRIX

*
* TO CHANGE AREA, MASS, GAMMA B OF SPACECRAFT
ARMARS (11.12)
MSMARS (259.00)
GSMARS (1.076)
ARMOON (2.789)
MSMOON (340.20)

```

JPL TECHNICAL MEMORANDUM NO. 33-204

```

G0K00N          (3.0)
ARVEN          (3.83)
MSVEN         (198.22)
GBVEN         (1.383)

*
* THE FOLLOWING ARE DOPPLER COEFFICIENTS AND
* FREQUENCIES WHICH ARE A FUNCTION OF THE
* BAND. NOMINALLY, STATIONS 1,2, AND 3 ARE
* S BAND, 4 AND 5 ARE L-S BAND.
*
*          L-BAND          L-S BAND          S BAND
TFREQ  TRANSPONDER FREQ  960.0E6        20.00E6        20.00E6
XFREQ  TRANSMITTER FREQ  29.66E6        2290.0E6       2290.0E6

DDP1(1)  (C1)   930.15E6   9.375E6   1.0E6
DDP1(2)  (C1)   -96875    .3125    1.0
DDP1(3)  (CC3)  -1E6      9.375E6  1.0E6
DDP1(4)  (CC3)  32.359559561  32.57918552  104.25339367
DDP1(5)  (C3)   930.15   9.375E6  1.0E6
DDP1(6)  (C3)  31.348314605  32.57918552  104.25339367

DDP2(1)  SAME AS ABOVE FOR STATION 2
DDP3(1)  SAME AS ABOVE FOR STATION 3
DDP4(1)  SAME AS ABOVE FOR STATION 4
DDP5(1)  SAME AS ABOVE FOR STATION 5

INFRQ  GROUND STATION INJECTION FREQUENCY, CPS (23.6279142E6)
SYNFRQ GROUND STATION SYNTHESIZER FREQUENCY, CPS (22.04209265E6)

L       FOLLOWED BY STATION ID TO ALTER STATION BAND
S       FOLLOWED BY STATION ID TO ALTER STATION BAND
LS      FOLLOWED BY STATION ID TO ALTER STATION BAND

ESTIMATE THESE PARAMETERS          (04)
CONSIDER THESE PARAMETERS          (05)

*
* THESE CARDS ARE FOLLOWED BY LISTS OF PARAMETER NAMES
* WHICH TELL THE DDP TO ESTIMATE (CONSIDER) THE
* CORRESPONDING PARAMETER. NO NUMERIC DATA REQUIRED.
* LIST OF PARAMETER NAMES WHICH CAN BE ESTIMATED
* (CONSIDERED).
*
* SYMBOLS ALLOWED
*
X,Y,Z,DX,DY,DZ,KE,RE,G,KM,MV,MM,MJ,J,H,D,AU,C,RI(1),LA(1),LO(1)
*
* OR
X1,Y1,Z1,DX1,DY1,DZ1,KE1,RE1,G1,KM1,MV1,MM1,FA1,J1,FB1,FC1,AU1,C1,RI1(1),LA1(1),LO1(1)

REJECTION SIGMAS                    (06)

*
* THE ABOVE CARD IS FOLLOWED BY DATA INDICATING WHICH OF
* THE OBSERVABLES ARE TO BE CHECKED FOR POSSIBLE REJECTION
* OF BAD POINTS. IF A SYMBOL IS ACCOMPANIED BY A NUMERICAL
* VALUE THE OBSERVATION WILL BE REJECTED IF THE ABSOLUTE
* VALUE OF THE RESIDUAL EXCEEDS THE INPUT VALUE.
*
* SYMBOLS ALLOWED

R(1),DR(1),EL(1),AZ(1),DEC(1),HA(1),C(1),CC3(1),
C3(1),DI(1),RU(1)

INVERSE COVARIANCE MATRIX OF ESTIMATED PARAMETERS (07)
COVARIANCE MATRIX OF ESTIMATED PARAMETERS (10)
COVARIANCE MATRIX OF CONSIDERED PARAMETERS (11)

*
* PROVISION IS MADE TO ENTER ONE OF THREE
* MATRICES IN ONE OF THREE WAYS
*
* TO ENTER THE DIAGONAL TERM OF THE MATRIX
DIAG=
*
* TO ENTER THE MATRIX BY ROW
R01=
*
* (CONTINUE ON FOLLOWING CARD, IF NECESSARY)
R02=
R03=
*
* TO ENTER WITH NO SYMBOLS SUPPLY A
* FORTRAN TYPE ARRAY FOR A 20X20 MATRIX

DELETE THESE DATA TYPES          (13)

*
* SYMBOLS ALLOWED AS IN 106 ABOVE

STATISTICS AND RESIDUALS FOR THESE DATA TYPES (14)

*
* DESIGNATION OF AT LEAST ONE DATA TYPE PER
* STATION IS NECESSARY FOR THE CALCULATION OF
* STATISTICS. INCLUDE ALL DATA TYPES FOR WHICH
* STATISTICS AND RESIDUALS ARE DESIRED. THE
* VERTICAL PLOTTING SCALE CAN BE DEFINED AS
* R(1)=10., THE HORIZONTAL SCALE CAN BE DEFINED
* AS NHR=1.5. THIS LATTER IS NOMINALLY 1. HOUR
* A MAXIMUM OF 8 DATA TYPES PER STATION MAY BE
* REQUESTED.
*
* SYMBOLS ALLOWED AS FOLLOWS

R(1),DR(1),EL(1),AZ(1),DEC(1),HA(1),C(1),CC3(1)
C3(1),DI(1),RU(1),NHR

NOMINAL VALUES CORRESPONDING TO COVARIANCE MATRIX (15)

*
* IF A COVARIANCE MATRIX FOR THE ESTIMATED PARAMETERS
* OR ITS INVERSE IS ENTERED THE VALUES OF THE
* ESTIMATED PARAMETERS ASSOCIATED WITH MATRIX MAY
* ALSO BE ENTERED. THESE VALUES FOLLOW THIS CONTROL
* CARD AND MUST BE IN THE SAME ORDER AS THE
* PARAMETERS IN THE MATRIX.
*
* IF NOMINAL VALUES FOR THE PARAMETERS ARE NOT
* ENTERED THE PROGRAM WILL SET THE NOMINAL
* VALUES TO THE INITIAL ESTIMATE OF THE PARAMETERS.
*
* NO SYMBOLS ARE ALLOWED.

```

WEIGHTS BY DATA TYPE AND STATION (116)

* NOMINALLY THE ODP COMPUTES THE WEIGHT FOR EACH DATA TYPE FROM INFORMATION THE ODP PUTS ON THE TRACKING DATA FILE. THE OPERATOR CAN OVERRIDE THIS FEATURE BY FOLLOWING THE ABOVE CARD WITH WEIGHTS (SIGMA RATHER THAN SIGMA SQUARED) FOR ANY OR ALL OF THE DATA TYPES. WEIGHTS MUST BE SPECIFIED IN THIS MANNER WHEN FITTING AN ODP-SIMULATED DATA FILE.

* THE USER MAY ALSO INPUT OCTAL WEIGHT CODES, E.G., CC3(2)=314333/8

* SYMBOLS ALLOWED

R(1),DR(1),EL(1),AZ(1),DEC(1),HA(1),CL(1),CC3(1)
C3(1),DI(1),RU(1)

POINTING TIMES, SAMPLE RATE, COUNT TIMES (117)

* AS THE NAME SUGGESTS, DATA FOLLOWING THIS CONTROL CARD DEFINES THE AMOUNT OF OUTPUT OBTAINED WHEN THE PROGRAM IS PREPARING A DATA FILE OR COMPUTING POINTING PREDICTIONS. A SYMBOLIC NAME IS ASSOCIATED WITH EACH OF THE 15 POSSIBLE TRACKING STATIONS, AND EACH NAME MAY BE FOLLOWED BY ONE, TWO, OR THREE GROUPS OF DATA. THE FORMAT OF EACH GROUP IS AS FOLLOWS: FIRST TIME, LAST TIME, SAMPLE RATE, COUNT TIME. POINTING PREDICTIONS (OR A DATA FILE) WILL BE GENERATED FOR THE INDICATED STATION IF THE PROBE IS ABOVE STATION'S HORIZON ANYTIME BETWEEN THE FIRST TIME AND THE LAST TIME. THE INTERVAL BETWEEN PREDICTIONS IS DEFINED BY THE SAMPLE RATE AND ANY REQUESTED DOPPLER CALCULATIONS WILL BE BASED ON THE INDICATED COUNT TIME. THE TWO TIMES ARE GREENWICH TIME AND ARE COMPOSED OF TWO WORDS EACH, AS DEFINED ELSEWHERE. THE SAMPLE RATE AND THE COUNT TIME ARE GIVEN AS FLOATING POINT SECONDS. AS AN EXAMPLE,

JETMTS=620702103,0,620702104,0,10,25,620702107,0,620702109,0,30,30.

* INSTRUCTS THE PROGRAM TO GENERATE POINTING PREDICTIONS FOR THE MTS EVERY 10 SEC BETWEEN 030000Z AND 040000Z ON JULY 21, 1962 WITH A DOPPLER AVERAGING TIME OF 5 SEC. POINTING PREDICTIONS WILL ALSO BE GENERATED FOR 070000Z TO 090000Z ON THE SAME DAY BUT WITH A 30-SEC SAMPLE RATE AND 30-SEC DOPPLER AVERAGING TIME.

* SYMBOLS ALLOWED

JETGLV,JETGL2,CANJET,DDHJET,JOBJET,ANTGUA,ASCENS,PUERTO,PRETRJ,JDDREL,T3CAPE,BAHAMA,SANSAL,HAWAII,TRINID

DATA FILE SIGMA (121)

* IF A DATA FILE IS BEING PREPARED THE DATA TYPES FOLLOWING THIS CARD WILL BE ENTERED ON THE ODP DATA FILE. IF NUMERIC FIELD APPEARS WITH A SYMBOL THE PROGRAM WILL PUT A NORMALLY DISTRIBUTED RANDOM NUMBER WITH MEAN ZERO ON THE CALCULATED VALUE OF THE DATA

* TYPE. THE ONE-SIGMA VALUE OF THE DISTRIBUTION IS SET TO THE VALUE FOLLOWING THE SYMBOLIC NAME OF THE DATA TYPE. IF NO NUMERIC FIELD FOLLOWS A SYMBOL THE ONE-SIGMA VALUE OF THE DISTRIBUTION IS SET TO ZERO.

* SYMBOLS ALLOWED

R(1),DR(1),EL(1),AZ(1),DEC(1),HA(1),CL(1),CC3(1)
C3(1),DI(1),RU(1)

DATA FILE BIAS (122)

* SAME AS THE ABOVE EXCEPT THAT THE NUMERIC FIELD INDICATES THE CONSTANT THE ODP WILL ADD TO THE CALCULATED VALUE OF EACH DATA TYPE.

PAGE HEADING (123)

* UP TO 11 BCD WORDS MAY FOLLOW THIS CONTROL CARD. THE COMMENT MUST BE INITIATED WITH A LEFT PARENTHESIS AND CLOSED WITH RIGHT PARENTHESIS. THE COMMENT WILL BE PRINTED ABOVE VARIOUS OUTPUT GROUPS.

MAP COVARIANCE MATRIX TO (124)

* THIS CARD IS FOLLOWED BY UP TO 12 MONOTONICALLY INCREASING GREENWICH TIMES. THE COVARIANCE MATRIX FROM THE PREVIOUS ITERATION WILL BE MAPPED FORWARD TO THESE TIMES.

* SYMBOLS ALLOWED, NONE

TRANSMITTER ID TABLE (126)

* THE DATA FOLLOWING THIS CONTROL CARD DEFINES THE TRANSMITTING STATION AS A FUNCTION OF TIME. THIS DATA IS REQUIRED ONLY WHEN GENERATING DATA FILES OR POINTING PREDICTIONS SINCE THE ODP IDENTIFIES THE TRANSMITTER FOR EACH TIME POINT APPEARING ON THE ODP DATA FILE. THE FORMAT OF THE TRANSMITTER TABLE IS, IX(1),FREQ(1),TIME(1),IX(2),FREQ(2),....IX IS A FIXED NUMBER, FREQ IS A FLOATING POINT NUMBER.

* THE INTERPRETATION OF THE TABLE IS AS FOLLOWS:

* STATION IX(1) IS TRANSMITTING WITH A FREQ(1) UNTIL TIME(1), THEN STATION IX(1)+1 IS TRANSMITTING WITH A FREQ(1)+1 UNTIL TIME(1)+1, ETC. IX(1)=0 IS INTERPRETED AS THE TRANSPONDER AND TIME(1)=0 IS INTERPRETED AS INFINITY. WHEN GENERATING A DATA FILE ONLY ONE TYPE OF DOPPLER WILL BE ALLOWED PER TIME POINT. IF ALL THREE DOPPLER TYPES ARE REQUESTED FOR A RECEIVING STATION (SEE DATA FILE SIGMA CONTROL CARDS), THE FOLLOWING RULES DETERMINE THE DOPPLER TYPE THAT WILL APPEAR.

* 1. IF ELEVATION AT STATION IX IS NEGATIVE SET IX=0

* 2. IF IX=0 OMIT CC3 AND C3

* 3. IF IX#0 OMIT C1 AND CC3 WHERE IR IS THE RECEIVER

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- * 4. IF IX=IR OMIT C1 AND C3
- * SYMBOLS ALLOWED, NONE
- * RESTRICTIONS
- * A MAXIMUM OF 134 IX-TIME PAIRS IS ALLOWED

OFFLINE CONTROL 127

- * WHEN ANY OPTION OTHER THAN A DATA FITTING
- * ITERATION IS DESIRED KEY CONTROL IS NECESSARY
- * THIS CONTROL CARD MUST BE PRESENT EACH ITERATION TO SET
- * THE KEYS FOR THAT ITERATION. THE CARDS FOR EACH
- * ITERATION MUST BE FOLLOWED BY AN END DATA CARD.
- * SYMBOL ALLOWED OPTION NEED CONTROL CARDS
- KEY(2) SIMULATE ODP DATA FILE 1,2,17,21
- KEY(3) STEP-MAP, FORWARDS EPOCH 1,2,24
- KEY(4) MAP FORWARD TO INPUT TIME 1,2,24
- KEY(5) MAP TO ENCOUNTER 1,2
- KEY(6) REJECT BAD POINTS 1,2,6
- KEY(8) REPEAT PREDICTS
- KEY(12) PREDICTS PREPARED FOR TELETYPE 1,2,17,KEY(13)
- TRANSMISSION
- KEY(13) POINTING PREDICTIONS 1,2,17
- KEY(14) STATISTICS 1,2,14
- KEY(15) ANGLE PLOTS 1,2
- KEY(16) RESIDUALS 1,2,14
- KEY(17) TIME PLOTS 1,2,14
- TERMINATE JOB MUST BE INCLUDED TO END JOB
- RESTART ODP TO INITIALIZE ODP FROM BEGINNING

BURN START TIME 130

- * THIS TIME, IN THE USUAL GREENWICH FORMAT, DENOTES THE
- * BEGINNING OF THE POWERED FLIGHT.

SPHERICAL INJECTION CONDITIONS 131

- * IN PLACE OF CARTESIAN INJECTION CONDITIONS, THE USER
- * MAY INPUT SPHERICAL CONDITIONS UNDER THIS CONTROL
- * CARD. THE SPHERICAL CONDITIONS ARE TRANSFORMED TO
- * CARTESIAN CONDITIONS AT INPUT TIME.
- * SYMBOLS ALLOWED
- RAD,LAT, LONG, VE, ELE, AZE

OCCULTATION TIMES 132

- * THIS DATA TYPE IS INPUT BY CARD, RATHER THAN THROUGH
- * THE JOB. TWO OCCULTATION TIMES APPEAR ON EACH CARD.
- * THE FORMAT IS OCCX=T1,T2,WEIGHT,REJECTION SIGMA, WHERE
- * T1 AND T2 ARE GREENWICH TIMES IN THE USUAL FORMAT.
- * RESTRICTIONS
- * ONLY STATIONS 1 THROUGH 5 PERMITTED. IF ONLY ONE
- * TIME IS TO BE USED THEN T2 MUST BE 0,0.
- * SYMBOLS ALLOWED

OCC1,OCC2,OCC3,OCC4,OCC5

IMPACT TIMES 133

- * SAME AS OCCULTATION TIME, EXCEPT THAT ONLY ONE TIME
- * PER STATION IS INPUT. THE FORMAT IS IMPX=T,WEIGHT,
- * REJECTION SIGMA.
- * RESTRICTIONS
- * ONLY STATIONS 1 THROUGH 5 PERMITTED
- * SYMBOLS ALLOWED
- IMP1,IMP2,IMP3,IMP4,IMP5

START AND STOP TIMES 134

- * THIS INPUT PERMITS THE FITTING OF DATA ONLY DURING THE
- * INTERVAL BETWEEN T-START AND T-STOP. UP TO THREE PAIRS
- * MAY BE INPUT IN THE USUAL GMT FORMAT.

END DATA 135

- * THIS CARD MUST TERMINATE EACH SET OF CONTROL DATA
- * READ BY THE ODP.

B. LOAD MAPS AND COMMON STORAGE MAP

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ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY.						
(DFMP)	(DFAD)	(DFSB)	READS	WRITES		
THE NAME OF THIS PROGRAM IS *LA2 2/22/65						
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.	OCTAL LENGTH	DECIMAL LENGTH	COMMON BREAK
**PCB	22302	EPDX	22300	00011	00009	
CDMAP	22322	CDMAP	22311	00273	00187	
		OFFSYS				
		TYPRYT				
		TAMP				
		PRDM				
		FLOT				
		REAM				
		COMHP				
		DCP				
		FLAX				
PRAMS	22606	DCP	22604	00043	00035	
COMHP	22660	FLAX	22647	01357	00751	47055
		PRAMS				
		(DFMP)				
		(DFAD)				
		MANUL				
		(DFSB)				
ADD	24226	*NONE*	24226	00050	00040	
ARCSIN	24306	*NONE*	24276	00144	00100	77152
ARCCOS	24302					
QARCSIN	24300					
QARCCOS	24276					
ERRARS	24427					
ERRARC	24433					
ARTAN	24442	*NONE*	24442	00103	00067	77152
DAYS	24550	FIX	24545	00041	00033	
		FLOT				
		ADD				
ERRDR	24606	*NONE*	24606	00011	00009	
FIXT	24620	FIXT	24617	00007	00007	
FLOT	24626	*NONE*	24626	00312	00202	50031
FIX	24734					
FLOT	25115					
DAYE	25121					
FLAT	25131					
FLAPR	25142	ERRDR	25140	00102	00066	
FLAK	25170	TYPRYT				
INSPEC	25246	*NONE*	25242	00310	00200	50031
MANUL	25556	*NONE*	25552	00130	00088	77461
PNUT	25710	SIN	25702	00336	00222	50001
		COS				
PRDM	26244	*NONE*	26240	00156	00110	77461
QUIZ	26416	*NONE*	26416	00003	00003	
REAP	26424	READS	26421	02543	01379	
REAM	26424	FLAT				
		ERRDR				
ROT	31170	*NONE*	31164	00266	00182	50031
OFFSYS	31452	*NONE*	31452	00010	00008	
ENDSYS	31453					
FINSYS	31454					
SCHAIN	31455					
PGSTRAT	31456					
PGSTOP	31457					
DCP	31460					
TYPRYT	31461					
SIN	31462	*NONE*	31462	00242	00162	
COS	31465					
QSIN	31471					
QCOS	31473					
TAMP	31736	WRITES	31724	00126	00086	
EPDX	32032	FLAT				
(DFAD)	32052	*NONE*	32052	00121	00081	77776
(DFSB)	32072					
(DFMP)	32112					
(DFDP)	32140					
READD	32174	(IOU)	32173	00365	00245	
READS	32176					
WRITES	32201					
WRITES	32203					
BSREC	32412					
BSFILE	32415					
REWIND	32420					
UNLOAD	32423					
EMOFILE	32426					
SETLBN	32404					
SETHI	32407					
(UNIT)	32555					
TAPEIO	32555					
(IOU)	32563	*NONE*	32560	00030	00024	
*LA2 * JUST LOADED.						
UNUSED CORE LIES FROM 32610 THROUGH 47055, LEAVING 14246 OCTAL OR 05310 DECIMAL LOCATIONS.						

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ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY,	(DFAD)	(DFAD)	SQRT	FGDDUT	PROUT	WRITED			
ATTACH ENDFIL	ATTACH REMINO	OPEN READB	CLOSE IDFSB)	(DFMP)	(DFAD)	SQRT	FGDDUT	PROUT	WRITED
THE NAME OF THIS PROGRAM IS 'LAZA ' 2/22/65									
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.	OCTAL LENGTH	DECIMAL LENGTH	COMMON BREAK			
**PCB NAPOUT	22324	NAPOUT DEFINE ATTACH OPEN TYPRYT OFFSYS CLOSE DCP FLAK PRAM PRIM REFORM TACCOM (DFMP)	22310	01564	03884				
AAT	24202		24074	00116	00078	77461			
BIBCO	24237	'NONE'	24212	00055	00045				
DIAGO	24274	NAPOUT	24267	00105	00089	50031			
ERROR	24374	'NONE'	24374	00011	00009				
FIXT	24406	FIXTI	24405	00007	00007				
FIXTI	24414	'NONE'	24414	00312	00202	50031			
PLOT	24522								
FIX	24703								
FLQAT	24707								
GATVE	24717								
FLAT	24730	ERRDR	24726	00101	00065				
FLAPR	24756	TYPRYT							
FLAK	24765								
NORMAY	25034	SQRT	25027	00251	00169	77461			
PRIM	25321	FGDDUT PRUII	25300	01423	00787	50001			
PRAM	25310	BIBCO FLAPR PRIME NORMAY PRIT DIAGO WRITEB							
PRIME	26730	FLAT	26723	00177	00127				
REFORM	26751	ENDFIL REMINO READB							
PRIT	27124	FIXT	27122	00036	00030				
QUIZ	27160	BIBCO	27160	00002	00002				
OFFSYS	27162	'NONE'	27162	00010	00008				
ENDSYS	27163								
FINSYS	27164								
SCHAIN	27165								
PGSTRT	27166								
PCSTOP	27167								
DCP	27170								
TYPRYT	27171								
TAPR	27172	'NONE'	27172	00002	00002				
SCALE	27200	'NONE'	27174	00043	00035	77461			
TACCOM	27250	SCALE (DFMP) (DFAD) AAT (DFSB) 'NONE'	27237	01617	00911	47055			
IDFAD)	31056		31056	00120	00080	77776			
IDFSB)	31076								
(DFMP)	31116								
(DFMP)	31144								
SQRT	31176	'NONE'	31176	00054	00044	77773			
PROUT	31266	OUTUS	31252	03631	01561				
FGDDUT	32532	ACTIND							
PACHV	31433	DFLC							
PRDUH2	31404	RESTKA							
PRDUH3	31417	REQIND							
PSKA	33441	WRITE							
TSXB	34231	PRCON RGGSAV RGGSTR CKIND CKACT (IOU)							
READD	34304		34303	00365	00245				
READB	34306								
WRITED	34311								
WRITEB	34313								
BSREC	34522								
BSFILE	34525								
REMINO	34530								
UNLOAD	34533								
ENDFIL	34536								
SETLOW	34514								
SETHI	34517								
(UNIT)	34665								
TAPETO	34665								
(IOU)	34673	'NONE'	34670	00030	00024				
OUTUS	34722	RGGSAV	34720	02320	01232				
BFLG	35556	RGGSTR							
ENDDUT	35307								
CKIND	37151								
CKACT	37214								
REQIND	37150								
ACTIND	37147								
RESTKA	37145								
PL1CON	37230								
PL2CON	37231								
PL3CON	37232								
PRCON	37233								
RGGSAV	37240	'NONE'	37240	00133	00091				
RGGSTR	37522								
IOCS	37374	CLOCK	37373	04335	02269				
OEFLINE	37375								
JOIN	37400								
ATTACH	37403								
CLOSE	37406								
OPEN	37411								
READ	37414								
WRITE	37417								
COPY	37422								
REW	37425								
WCF	37430								
MSR	37433								
BSF	37436								

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STASH 37441
 CLOCK 43733
 MINUTE 43730
 XMIN 43730

'LAZA ' JUST LOADED.

UNUSED CORE LIES FROM 44036 THROUGH 47055, LEAVING 03020 OCTAL OR 01552 DECIMAL LOCATIONS.

ENTRY POINTS TO DEFINE	SUBROUTINES ATTACH	REQUESTED FROM LIBRARY, OPEN	CLOSE	WRITEB	ENDFIL	FGDOUT	SQRT	PROUT	LOC
THE NAME OF THIS PROGRAM IS 'LA3 ' 2/22/65									
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.		OCTAL LENGTH		DECIMAL LENGTH	COMMON BREAK	
*PCB	22302	OZ	22301		00006		00006		
OZ	22316	DEFINE ATTACH OPEN TYPRYT OFFSYS UMAT CLOSE	22307		01367		00759		
ARTAN	23676	'NONE'	23676		00103		00067	77152	
FIXT	24002	FIXTT	24001		00007		00007		
FIXTT	24010	'NONE'	24010		00312		00202	50031	
FLOT	24116								
FIX	24277								
FLOAT	24303								
DAYE	24313								
ERROR	24322	'NONE'	24322		00012		00010		
FLAT	24336	ERROR	24334		00101		00065		
FLAPR	24364	TYPRYT							
FLAK	24373								
GERPU	24440	WRITEB FLAT	24435		00100		00064		
ORBEQ	24544	ENDFIL ARTAN COS	24535		00167		00119	77461	
PARAM	24732	SIN FGDOUT SORT FIXT PROUT INORM ORBEQ	24724		01320		00720		
QUIZ	26244	'NONE'	26244		00002		00002		
OFFSYS	26246	'NONE'	26246		00010		00008		
ENDSYS	26247								
FINSYS	26250								
SCHAIN	26251								
PGSTRP	26252								
PGSTDP	26253								
DCP	26254								
TYPRYT	26255								
SIN	26256	'NONE'	26256		00243		00163		
COS	26261								
QSIN	26265								
QCOS	26267								
INORM	26526	SORT	26521		00252		00170	77461	
UMAT	27004	SORT SORT LOG GERPU ARTAN PARAM	26773		03221		01681	77461	
LOG	32222	'NONE'	32214		00066		00054	77774	
LOG10	32214								
SORT	32302	'NONE'	32302		00054		00044	77773	
PROUT	32332	OUTS	32356		03031		01561		
FGDOUT	33636	ACTIND							
PRCON	32537	BFLG							
PROUT2	32510	RESTKA							
PROUT3	32523	REQIND							
TSXA	34545	WRITE							
TSXB	35335	PRCON RGGSAV RGGSTR CKACT IDOU)							
READD	35410		35407		00365		00245		
READ8	35412								
WRITED	35415								
WRITEB	35417								
BSREC	35626								
BSFILE	35631								
REWIND	35634								
UNLOAD	35637								
ENDFIL	35642								
SETLOW	35620								
SETHI	35623								
(UNIT)	35771								
TAPEIO	35771								
IDOU)	35777	'NONE'	35774		00030		00024		
OUTUS	36026	RGGSAV	36024		02320		01232		
BFLG	36662	RGGSTR							
ENDOUT	36413								
CKIND	40295								
CKACT	40320								
REQIND	40254								
ACTIND	40253								
RESTKA	40251								
PLICON	40334								
PL2CON	40335								
PL3CON	40336								
PRCON	40337								
AGGSAV	40344	'NONE'	40344		00133		00091		
RGGSTR	40426								
IDCS	40500	CLOCK	40477		04335		02269		
DEFINE	40501								
JOIN	40504								
ATTACH	40507								

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CLOSE      40512
OPEN       40515
READ       40520
WRITE      40523
COPY       40526
REW        40531
WEP        40534
BSR        40537
BSF        40542
STASH      40545
CLOCK      45037      *NONE*      45034      00106      00070
MINUTE     45034
XMIN       45034
    
```

*LA3 * JUST LOADED.

UNUSED CORE LIES FROM 45142 THROUGH 50031, LEAVING 02670 OCTAL OR 01464 DECIMAL LOCATIONS.

ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY,	REMINO	(DFAD)	(DFMP)	(DFSB)	SQRT	DMDD	(DFOP)	EXP(3)	READB
THE NAME OF THIS PROGRAM IS *LA4 *									
2/22/65									
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.	OCTAL LENGTH	DECIMAL LENGTH	COMMON BREAK			
*PCB	22302	PRED	22301	00007	00007				
PRED	22314	TYPRY	22310	00050	00040				
		POINT							
		DATAPE							
ABOD	22360	*NONE*	22360	00030	00024				
ADD	22410	*NONE*	22410	00050	00040				
ARSIN	22470	*NONE*	22460	00144	00100	77152			
ARCOS	22464								
QARSIN	22462								
QARCOS	22460								
ERRARC	22611								
ERRARC	22615								
ARTAN	22624	*NONE*	22624	00102	00066	77152			
COEF	22736	(DFAD)	22726	00770	00504	50031			
		(DFMP)							
		(DFSB)							
		QUIZ							
CORR	23724	SQRT	23716	00274	00188	47675			
		SIN							
DATAPE	24230	IXTAB	24212	00447	00295	50031			
		OFFSYS							
		(DFAD)							
		REAP							
		GHADP							
		COEF							
		DDPLR							
		CORR							
		NDZF							
		FILT							
DAYS	24664	FIX	24661	00041	00033				
		FLOAT							
		ADD							
DNAME	24735	*NONE*	24722	00014	00012				
SIPREG	24736	*NONE*	24736	00003	00003				
ODOFF	24737								
SPHX	24740								
DDPLR	24756	RATES	24741	01313	00715	50031			
		DMDD							
		VEC							
		ABOD							
		(DFMP)							
		(DFAD)							
		SORT							
		ARTAN							
		ARSIN							
ERROR	26254	*NONE*	26254	00012	00010				
FILL	26270	RITE	26266	00277	00191	50001			
FILT	26552	PISA							
FIXT	26566	FIXTT	26565	00007	00007				
FIXIT	26574	*NONE*	26574	00312	00202	50031			
FLOT	26702								
FIX	27063								
FLOAT	27067								
DAYE	27077								
FLAT	27110	ERRDR	27106	00102	00066				
FLAPR	27136	TYPRY							
FLAK	27145								
GHADP	27222	(DFOP)	27210	00320	00208	50031			
		(DFSB)							
		(DFMP)							
		(DFAD)							
		DMDD							
		COS							
IXTAB	27530	*NONE*	27530	00066	00054				
NDZF	27616	*NONE*	27616	00050	00040				
PNUT	27674	SIN	27666	00337	00223	50001			
		COS							
POINT	30242	IXTAB	30225	00305	00197	50031			
		COEF							
		OFFSYS							
		(DFAD)							
		REAP							
		GHADP							
		DDPLR							
		CORR							
		FILL							
QUIZ	30532	*NONE*	30532	00003	00003				
RATES	30546	COS	30535	00662	00434	50031			
		SIN							
		SQRT							
		ARSIN							
REAP	31422	EXP(3)	31417	02543	01379				
REAM	31422	READB							
		FLAT							
		ERROR							
RDT	34166	*NONE*	34162	00266	00182	50031			
OFFSYS	34450	*NONE*	34450	00010	00008				
ENDSYS	34451								
FINSYS	34452								
SENATH	34453								

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PGSTRT	34454					
PGSTOP	34455					
DCP	34456					
TYPRYT	34457					
SIN	34460	'NONE'	34460	00242	00162	
COS	34463					
QSIN	34467					
QCOS	34471					
VEC	34722	'NONE'	34722	00017	00015	50031
REDE	34742	DCP	34741	00421	00273	
RITE	34745					
PISA	35171					
EXPI3	35362	'NONE'	35362	00136	00094	77773
(OFAD)	35520	'NONE'	35520	00120	00080	77776
(OFSB)	35540					
(DFMP)	35560					
(OFDP)	35506					
OMD	35640	'NONE'	35640	00060	00048	77776
SQRT	35720	'NONE'	35720	00059	00045	77773
READD	35776	(IOU)	35775	00365	00245	
READB	36000					
WRITED	36003					
WRITEB	36009					
BSREC	36214					
BSFILE	36217					
REWIND	36222					
UNLOAD	36225					
ENDFIL	36230					
SELDW	36206					
SETH	36211					
(UNIT)	36357					
TAPEIO	36357					
(IOU)	36365	'NONE'	36362	00030	00024	

'LA4 * JUST LOADED.

UNUSED CORE LIES FROM 36412 THROUGH 47675, LEAVING 11264 OCTAL OR 04788 DECIMAL LOCATIONS.

ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY,							
DEFINE	ATTACH	OPEN	CLOSE	PROUT	FGDDUT	TELTYP	ENDOUT
THE NAME OF THIS PROGRAM IS 'LA5 * 2/22/65							
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.	OCTAL LENGTH	DECIMAL LENGTH	COMMON BREAK	
**PCB	22302	SORT	22301	00006	00006		
SORT	22316	TYPRYT	22307	02055	01069		
		DEFINE					
		ATTACH					
		OPEN					
		THARP					
		CLOSE					
		FLAK					
BIBCO	24411	'NONE'	24364	00054	00044		
DNAME	24453	'NONE'	24440	00014	00012		
AQUI	24512	PROUT	24454	00741	00481	50031	
ALLI	24472	FGDDUT					
BALLI	24467	PKINE					
BAQUI	24464	FIXT					
		DAYE					
		TELTYP					
		ENDOUT					
PERNOD	25421	BIBCO					
WOLF	25420	KINE	25415	00007	00007		
RESP	25416						
ANGPLT	25417						
MONRED	25424	'NONE'	25424	00002	00002		
ERRDR	25426	'NONE'	25426	00012	00010		
FILL	25442	RITE	25440	00277	00191	50001	
FILT	25724	PISA					
FIXT	25740	FIXTT	25737	00007	00007		
FIXTT	25746	'NONE'	25746	00312	00202	50031	
FLOT	26054						
FIX	26235						
FLOAT	26241						
DAYE	26251						
FLAT	26262	ERRDR	26260	00102	00066		
FLAPR	26310	TYPRYT					
FLAK	26317						
KINE	26364	REDE	26362	00216	00142	50001	
QUIT	26400	ERRDR	26400	00002	00002		
OFFSYS	26402	'NONE'	26602	00010	00008		
ENDSYS	26603						
FINSYS	26604						
SCHATT	26605						
PGSTRT	26606						
PGSTOP	26607						
UCP	26610						
TYPRYT	26611						
THARP	26626	OFFSYS	26612	00573	00379		
PKINE	26733	KINE					
		AQUI					
		ALLI					
		BAQUI					
		BALLI					
		PERNOD					
		RESP					
		WOLF					

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REDE	27406	ANGPLT	27405	00422	00274
RITE	27411	BIBCD			
PISA	27635	TYPRYT			
TELTYP	30036	DCP			
SCCTTY	30253	PROUT2	30027	01101	00577
		ACTIND			
		PROUT3			
		TSXA			
		RGCSAV			
		RGGSTR			
		TSXB			
PROUT	31144	OUTUS	31130	03032	01562
FGDOUT	32410	ACTIND			
PRGNV	31311	BFLC			
PROUT2	31262	RESTKA			
PROUT3	31275	REQIND			
TSXA	33317	WRITE			
TSXB	34107	PRCON			
		RGCSAV			
		RGGSTR			
		CKIND			
		CACT			
		RGCSAV	34162	02320	01232
		RGGSTR			
OUTUS	34164				
BFLC	35020				
ENDOUT	34551				
CKIND	36413				
CACT	36456				
REQIND	36412				
ACTIND	36411				
RESTKA	36407				
PLICDN	36472				
PL2CDN	36473				
PL3CDN	36474				
PRCON	36475				
RGCSAV	36502	*NONE*	36502	00133	00091
RGGSTR	36564				
ICCS	36636	CLOCK	36635	04335	02269
DEFINE	36637				
JOIN	36642				
ATTACH	36645				
CLOSE	36650				
OPEN	36653				
READ	36656				
WRITE	36661				
COPY	36664				
REW	36667				
HEF	36672				
BSR	36675				
BSF	36700				
STASH	36703				
CLOCK	43175	*NONE*	43172	00106	00070
MINUTE	43172				
AMIN	43172				

*LA5 * JUST LOADED.

UNUSED CORE LIES FROM 43300 THROUGH 50001, LEAVING 04502 OCTAL OR 02370 DECIMAL LOCATIONS.

ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY,	SETHI	REWIND	ENDFIL	(DFSB)	(DFAD)	(DFNP)	SQRT	EXPI3	DNDD	(DFDP)
THE NAME OF THIS PROGRAM IS	*LA6 *			2/22/65						
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.	OCTAL LENGTH	DECIMAL LENGTH	COMMON BREAK				
**PCB	22302	FIT	22301	00006	00006					
FIT	22320	REDE	22307	00735	00477					
		TYPRYT								
		SETHI								
		REWIND								
		CALL								
		ENDFIL								
		RITE								
		OCIM								
		FLAK								
ABDD	23244	*NONE*	23244	00030	00024					
ADD	23274	*NONE*	23274	00050	00040					
ARSHN	23354	*NONE*	23344	00144	00100	77152				
ARCOS	23350									
QARSHN	23346									
QARCOS	23344									
ERRARS	23475									
ERRARC	23501									
ARTAN	23510	*NONE*	23510	00103	00067	77152				
BEER	23614	REJEC	23613	00071	00057	47675				
CALL	23736	BEER	23704	00523	00339	51235				
		KINE								
		COEF								
		STARP								
		(DFSB)								
		REAP								
		GHADP								
		DDPLK								
		COROP								
		CORR								
		DICDS								
		DBTOX								
		CATS								
		OFFSYS								
		WAIT								
		QUIZ								
		REJEC								
		PM360								
		FORM								
		CERTA								
		FILL								
		GREDF								
CATS	24436	VEC	24427	00645	00421	47675				
		ABDD								
		CDS								
		(DFAD)								
		(DFNP)								
		(DFSB)								
		QUIZ								
		CDS								
		SQRT								
		EXPI3								
		ARTAN								
COEF	25304		25274	00770	00504	50031				
CDL	26272		26264	00406	00262	50031				
COROP	26702		26672	00170	00120	50031				

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CORR	27070	SIN	27062	00275	00189	47675
DAYS	27362	SORT	27357	00041	00033	
DECDD	27420	SIN	27420	00040	00032	50031
DICOS	27466	SIN	27460	00173	00123	50031
DOPLR	27670	COS	27653	01313	00715	50031
		RATES				
		DMOD				
		VEC				
		ARDD				
		(DFMP)				
		(DFAD)				
		SQRT				
		ARTAN				
		ARSIN				
ERROR	31166	*NONE*	31166	00012	00010	
FILL	31202	RITE	31200	00277	00191	50001
FILT	31464	PISA				
FIXT	31500	FIXTT	31477	00007	00007	
FIXTT	31506	*NONE*	31506	00312	00202	50031
FLOT	31614					
FIX	31775					
FLOAT	32001					
DAYE	32011					
FLAT	32022	ERROR	32020	00101	00065	
FLAPR	32050	TYPRYT				
FLAK	32057					
FORM	32132	PARLEY	32121	01066	00566	50031
		(DFMP)				
		(DFAD)				
		(DFOP)				
		SORT				
		WRITB	33207	00215	00141	
GERTA	33212	FLAT				
GRODF	33332	ENDFIL				
		(DFDP)	33424	00321	00209	50031
		(DFSB)				
		(DFMP)				
		(DFAD)				
		DMOD				
		COS				
		FORM	33745	00177	00127	50031
IMPAR	33752	*NONE*	34144	00310	00200	50031
INSPC	34150	REDE	34454	00216	00142	50001
KINE	34456	ERRDR				
		NONE	34672	00275	00189	50031
DBTOX	34676	IMPAR	35167	00222	00146	
OCIM	35174	DCPAR				
		READB				
		FLAT				
		ERROR	35411	00515	00333	50031
DCPAR	35416	FORM	36126	00044	00036	
PARLEY	36130	WRITB				
		FLAT				
		NONE	36172	00050	00040	50001
PH360	36176	SIN	36242	00336	00222	50001
PHUI	36250	COS				
		NONE	36600	00003	00003	
QUIZ	36600	COS	36603	00662	00434	50031
RATES	36614	SIN				
		SORT				
		ARSIN				
		EXPL3				
REAP	37470	READB	37465	02543	01379	
REAN	37470	FLAT				
		ERROR				
		NONE	42230	00056	00046	46711
REJEC	42230					
RECO	42302					
ROT	42312	*NONE*	42306	00266	00182	50031
OFFSYS	42574	*NONE*	42574	00010	00008	
ENDSYS	42575					
FINSYS	42576					
SCHATTN	42577					
PGSTRT	42600					
PGSTOP	42601					
DCP	42602					
TYPRYT	42603					
SIN	42604	*NONE*	42604	00242	00162	
COS	42607					
QSIN	42813					
OCOS	42615					
STARP	43046	*NONE*	43046	00030	00024	
VEC	43076	*NONE*	43076	00020	00016	50031
WAIT	43130	CUL	43116	00346	00230	50031
		DECDD				
		SORT				
		WDCT				
		COS				
		ERROR	43464	00013	00011	
WDCT	43464	*NONE*	43477	00421	00273	
REDE	43500	DCP				
RITE	43503					
PISA	43727					
EXPI3	44120	*NONE*	44120	00136	00094	77773
(DFAD)	44256	*NONE*	44256	00120	00080	77776
(DFSB)	44276					
(DFMP)	44316					
(DFDP)	44344					
DMOD	44376	*NONE*	44376	00050	00048	77776
SQRT	44456	*NONE*	44456	00055	00045	77773
READO	44534	(IDU)	44533	00365	00245	
READB	44536					
WRITB	44541					
WRITB	44543					
BSREC	44752					
BSFILE	44755					
REWIND	44760					
UNLOAD	44763					
ENDFIL	44766					
SETLOW	44784					
SETHI	44747					
(UNIT)	45115					
TAPEIO	45115					
(IDU)	45123	*NONE*	45120	00030	00024	

*LA6 * JUST LOADED.

UNUSED CORE LIES FROM 45150 THROUGH 46711. LEAVING 01542 OCTAL OR 00866 DECIMAL LOCATIONS.

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ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY	ATTACH	OPEN	CLOSE	FGDDUT	PROUT	SURT	IDFSB)	IDFMP)	IDFAD)
DEFINE CLOCK	(DFDP)								
THE NAME OF THIS PROGRAM IS 'LAT' 2/22/65									
ENTRY NAME **PCB	ENTRY ADD. 22302	TRANSFER VECTORS	LOAD ADD. 22300	DCIAL LENGTH 00012	DECIMAL LENGTH 00010	COMMON BREAK			
PRINT	22330	PRINT DEFINE ATTACH OPEN ENDIT RITE OFFSYS PRINQ PRIM CLOSE REDE FGDDUT FIXT PROUT FLAK	22312		02514	01356			
BIRCD	25053	*NONE*	25026	00055	00045				
DIAG	25110	SQRT	25103	00106	00070	50031			
DIAGO	25216	SQRT	25211	00105	00069	50031			
DNAME	25331	*NONE*	25316	00014	00012				
ENDIT	25346	(DFSB) (DFMP) (DFAD)	25332	01474	00828	50031			
		PRIN SQRT STPREG REVRT NDUT							
ERRUR	27026	*NONE*	27026	00011	00009				
FIXT	27040	FIXT	27037	00007	00007				
FIXIT	27046	*NONE*	27046	00912	00202	50031			
FLOT	27154								
FIX	27335								
FLOAT	27341								
DAYE	27351								
FLAT	27352								
FLAPR	27410	EARDR TYPRYT	27360	00101	00065				
FLAK	27417								
NORMAY	27456	SQRT	27461	00251	00169	77461			
NDUT	27734	PROUT	27732	00100	00064				
BLEW	27776	TYPRYT							
BUKEY	30032	*NONE*	30032	00040	00032				
OPKEY	30036								
COKEY	30053								
AFKEY	30060								
PRIM	30113	FGDDUT	30072	01424	00788	50001			
PRAN	30102	PROUT BIRCD FLAPR PRIME NORMAY PRIF DIAGO							
		NONE	31516	00005	00005				
TAMP	31517								
PRIME	31520								
TAPR	31521								
DIAGO	*****								
PRINQ	31532	FGDDUT	31523	00435	00285	50031			
PRIT	31763	DIAG CLOCK PROUT FLAPR FIXT BIRCD							
APRIOR	32154	*NONE*	32160	00002	00032				
		COKEY BLEW	32162	00206	00134	50031			
QUIZ	32160	*NONE*	32370	00010	00008				
REVRT	32170	*NONE*							
OFFSYS	32370								
ENDSYS	32371								
FINSYS	32372								
SCHAIN	32373								
PGSTRT	32374								
PGSTOP	32375								
DCP	32376								
TYPRYT	32377								
STPREG	32410	(DFAD) (DFDP) DSQRT (DFMP)	32400	02055	01069	71757			
DSQRT	34456	ERROR	34455	00106	00070	77776			
REDE	34564	OCF	34563	00421	00273				
RITE	34567								
PISA	35013								
(DFAD)	35204	*NONE*	35204	00120	00080	77776			
(DFSB)	35224								
(DFMP)	35244								
(DFDP)	35272								
SQRT	35324	*NONE*	35324	00054	00044	77773			
CLOCK	35403	*NONE*	35400	00106	00070				
MINUTE	35400								
XMIN	35400								
PROUT	35522	DUTUS	35506	03032	01562				
FGDDUT	36766	ACTIND							
PRCNV	36667	BFLG							
PROUTZ	36640	RESTKA							
PROUT3	36653	REQIND							
TSXA	37675	WRITE							
TSXB	40465	PRCON RGGSAV RGGSTR CKIND CKACT							
		RGGSAV RGGSTR	40540	02320	01232				
DUTUS	40542								
BFLG	41376								
ENDOUT	41127								
CKIND	42771								
CKACT	43034								
REQIND	42770								
ACTIND	42767								
RESTKA	42765								
PLICON	43050								
PLCON	43051								
PLCON	43052								
PRCON	43053								
RGGSAV	43060	*NONE*	43060	00133	00091				
RGGSTR	43142								
IOCS	43214	CLOCK	43213	04335	02269				
DEFINE	43215								
JOIN	43220								
ATTACH	43223								

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CLOSE 43226
 OPEN 43231
 READ 43234
 WRITE 43237
 COPY 43242
 REM 43245
 WEF 43250
 BSR 43253
 OSP 43256
 STASH 43261

*LA7 * JUST LOADED.

UNUSED CORE LIES FROM 47550 THROUGH 50001, LEAVING 00232 OCTAL OR 00154 DECIMAL LOCATIONS.

ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY	DEFINE	ATTACH	OPEN	CLOSE	WEF	WRITE	FGDDUT	PROUT	SORT
THE NAME OF THIS PROGRAM IS 'LAB ' 2/22/65									
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.			OCTAL LENGTH	DECIMAL LENGTH	CDMMDN BREAK	
**PCB	22302	RESID	22301			00007	00007		
CLOCK	22310	*NONE*	22310			00002	00002		
RESID	22320	DEFINE	22312			01301	00705		
		ATTACH							
		OPEN							
		TYPRYT							
		THARP							
		CLOSE							
ANGPLT	23520	PKINE	23613			00577	00383	51235	
		FIXT							
		BTBCD							
		CXPLOT							
		BTBCD	24412			00054	00044		
		CXPLOT	24470			03320	01744		
		XXXXSR	25327						
		CDC	25575						
		PLTBNE	24521						
		SET	24472						
		DIFD	30006	*NONE*	30006	00026	00022		
		DNAM	30047	*NONE*	30034	00015	00013		
		BALLI	30056	XINE	30051	00007	00007		
		BAQUI	30057						
		ALLI	30052						
		AQUI	30053						
		JUDY	30054						
		ERRDR	30060	*NONE*	30060	00011	00009		
		FIXT	30072	FIXT	30071	00007	00007		
		FIXT	30100	*NONE*	30100	00312	00202	50031	
		FLOT	30206						
		FIX	30367						
		FLOAT	30373						
		DAYE	30403						
		FLAT	30414	ERRDR	30412	00102	00066		
		FLAPR	30442	TYPAYT					
		FLAK	30451						
		XINE	30516	REDE	30514	00215	00141	50001	
		ERRDR		ERRDR					
		PERNOD	30736	FIXT	30731	00503	00323	51235	
		WOLF	31067	FGDDUT					
				PROUT					
				SORT					
				COLA					
				NONE	31434	00002	00002		
				CDC	31436	01564	00884	50001	
				FGDDUT					
				CXPLOT					
				PKINE					
				PERNOD					
				FIXT					
				PROUT					
				FLAPR					
				BTBCD					
				ERRDR					
				NONE	33222	00010	00008		
OFFSYS	33222								
ENDSYS	33223								
FINSYS	33224								
SCHAIN	33225								
PGSTRT	33226								
PGSTOP	33227								
DCP	33230								
TYPRYT	33231								
STATID	33232	*NONE*	33232			00020	00016		
THARP	33266	OFFSYS	33252			00573	00379		
PKINE	33373	XINE							
		AQUI							
		ALLI							
		BAQUI							
		BALLI							
		PERNOD							
		RESP							
		WOLF							
		ANGPLT							
		BTBCD							
		TYPRYT							
		DCP							
REDE	34046		34045			00421	00273		
RTIE	34051								
PISA	34275								
SORT	34466	*NONE*	34466			00054	00044	77773	
PROUT	34556	OUTUS	34542			03032	01562		
FGDDUT	36022	ACTIND							
PRCNV	34723	BFLG							
PROUT2	34674	RESTKA							
PROUT3	34707	REQIND							
TSXA	34731	NAIIE							
TSXB	37521	PRCON							
		RGGSAY							
		RGGSTR							
		CKIND							
		CKACT							
		RGGSAY	37574			02320	01232		
		RGGSTR							
OUTUS	37576								
BFLG	40432								
ENDOUT	40163								
CKIND	42025								
CKACT	42070								
REQIND	42024								
ACTIND	42023								

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RESTKA 42021
PL1CON 42104
PL2CON 42105
PL3CON 42106
PRCON 42107
RGSAY 42114 *NONE* 42114 00133 00091
RGSSTR 42176
IGCS 42250 CLOCK 42247 04335 02269
DEFINE 42251
JDIN 42254
ATTACH 42257
CLOSE 42262
OPEN 42265
READ 42270
WRITE 42273
COPY 42276
REW 42301
WEF 42304
BSR 42307
BSF 42312
STASH 42315
    
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*LAB * JUST LOADED.

UNUSED CORE LIES FROM 46604 THROUGH 50001, LEAVING 01176 OCTAL OR 00638 DECIMAL LOCATIONS.

ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY,									
REWIND	ENDFIL	SQRT	(DFDP)	(DFS8)	(DFMP)	(DFAD)	DMDD	DEFINE	ATTACH
OPEN	CLOSE	EXPI3	RGSAY	RGSSTR	CLOCK	READ			
THE NAME OF THIS PROGRAM IS *LA9 * 2/22/65									
ENTRY NAME	ENTRY ADD.	TRANSFER	VECTORS	LOAD ADD.	OCTAL LENGTH	DECIMAL LENGTH	COMMON	BREAK	
MOOPH1	22460	TYPRYT		22300	01322	00722			
ODOFF	22414	ODATA							
READS	22306	REMINO							
		OFFSYS							
		ENDFIL							
		ERRDR							
TELTYP	23624	*NONE*		23622	00005	00003			
TELTP	23623								
IUNIT)	23622								
FIXT	23526	FIXIT		23625	00010	00008			
DIAG	23642	SQRT		23635	00105	00069			50031
ERRDR	23744	TYPRYT		23742	00442	00290			
		ONLIN							
FIXTT	24404	*NONE*		24404	00312	00202			50031
FLOT	24512								
FIX	24673								
FLOAT	24677								
DAYE	24707								
GHADP	24730	(DFDP)		24716	00320	00208			50031
		(DFS8)							
		(DFMP)							
		(DFAD)							
		DMDD							
		COS							
QUIZ	25236	*NONE*		25236	00002	00002			
OFFSYS	25240	*NONE*		25240	00010	00008			
ENDSYS	25241								
FINSYS	25242								
SCHATN	25243								
PGSTRT	25244								
PGSTOP	25245								
DCP	25246								
TYPRYT	25247								
SIN	25250	*NONE*		25250	00243	00163			
COS	25253								
QSIN	25257								
QCOS	25261								
SPHX	25522	COS		25513	00323	00211			50031
		SIN							
		GHADP							
STPREG	26046	(DFAD)		26036	02055	01069			71757
		(DFDP)							
		OSORT							
		(DFMP)							
DSQRT	30114	ERRDR		30113	00106	00070			77776
REDE	30222	DCP		30221	00422	00274			
RITE	30225								
PISA	30451								
ODATA	30566	TYPRYT		30643	03305	01733			50001
ONLIN	30770	DEFINE							
*****	31136	ATTACH							
		OPEN							
		CARDS							
		OFFSYS							
		ERRDR							
		FINSYS							
		CLOSE							
		ODOFF							
		READS							
		MOOPH1							
		FLOT							
		SPHX							
		STPREG							
		DIAG							
		REDE							
		DCP							
CARDS	34154	EXPI3		34150	02266	01206			
CROCON	34430	RGSAY							
CROCNV	34450	RGSSTR							
		CLOCK							
		READ							
EXPI3	36436	*NONE*		36436	00136	00094			77773
(DFAD)	36574	*NONE*		36574	00120	00080			77776
(DFS8)	36614								
(DFMP)	36634								
(DFDP)	36662								
DMDD	36714	*NONE*		36714	00060	00048			77776
SQRT	36774	*NONE*		36774	00054	00044			77773
CLOCK	37053	*NONE*		37050	00107	00071			
MINUTE	37050								
MIN	37050								
READD	37160	(IDU)		37157	00365	00245			
READD	37162								
WRITEO	37165								

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WRITB	37167				
BSREC	37376				
BSFILE	37401				
REWIND	37404				
UNLOAD	37407				
ENDFIL	37412				
SETLOW	37370				
SETHI	37373				
(UNIT)	**4444				
TAPEIO	37541				
(IOU)	37547	*NONE*	37544	00030	00024
RGGSAY	37574	*NONE*	37574	00133	00091
RGGSSTR	37656				
IOCS	37730	CLOCK	37727	04335	02269
DEFINE	37731				
JOIN	37734				
ATTACH	37737				
CLOSE	37742				
OPEN	37745				
READ	37750				
WRITE	37753				
COPY	37756				
REW	37761				
WEF	37764				
BSR	37767				
BSF	37772				
STASH	37775				

'LA9 * JUST LOADED.

UNUSED CORE LIES FROM 44264 THROUGH 50001, LEAVING 03516 OCTAL OR 01870 DECIMAL LOCATIONS.

ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY.

FGDDUT	DEFINE	ATTACH	OPEN	CLOSE	COMENT	READB	PROUT	
THE NAME OF THIS PROGRAM IS 'LA10 ' 2/22/65								
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.		OCTAL LENGTH	DECIMAL LENGTH	COMMON BREAK	
**PCB	22302	PARSH	22301		00006	00006		
PARSH	22316	FGDDUT	22307		01226	00662		
		DEFINE						
		ATTACH						
		OPEN						
		CONDUT						
		WASH						
		CLOSE						
COMOUT	23536	COMENT	23535		00016	00014		
FIXT	23554	FIXTT	23553		00007	00007		
FIXTT	23562	*NONE*	23562		00312	00202	50031	
FLOT	23670							
FIX	24051							
FLOAT	24055							
DAYE	24065							
FLAT	24076	ERROR	24074		00102	00066		
FLAPR	24124	TYPRYT						
FLAK	24133							
ERROR	24176	*NONE*	24176		00011	00009		
WASH	24214	READB	24207		00623	00403		
		FLAT						
		FIXT						
		PROUT						
		FLAPR						
OFFSYS	25032	*NONE*	25032		00010	00008		
ENDSYS	25033							
FINSYS	25034							
SCHAIN	25035							
PGSTRF	25036							
PGSTOP	25037							
DCP	25040							
TYPRYT	25041							
PROUT	25056	OUTUS	25042		03031	01561		
FGDDUT	26322	ACTIND						
PRCNY	25423	BFLC						
PROUT2	25174	RESTKA						
PROUT3	25207	REQIND						
TSXA	27231	WRITE						
TSXB	30021	PRCON						
		RGGSAY						
		RGGSSTR						
		CKIND						
		EXACT						
		(IOU)	30073		00365	00245		
READD	30074							
READB	30076							
WRITB	30101							
WRITB	30103							
BSREC	30312							
BSFILE	30315							
RFWIND	30320							
UNLOAD	30323							
ENDFIL	30326							
SETLOW	30304							
SETHI	30307							
(UNIT)	30355							
TAPEIO	30455							
(IOU)	30463	*NONE*	30460		00030	00024		
OUTUS	30512	RGGSAY	30510		02320	01232		
BFLC	31346	RGGSSTR						
ENDOUT	31977							
CKIND	32741							
CKACT	33004							
REQIND	32740							
ACTIND	32737							
RESTKA	32735							
PLICON	33020							
PL2CON	33021							
PL3CON	33022							
PRCON	33023							
RGGSAY	33030	*NONE*	33030		00133	00091		
RGGSSTR	33112							
IOCS	33164	CLOCK	33163		04336	02270		
DEFINE	33165							
JOIN	33170							
ATTACH	33173							
CLOSE	33176							
OPEN	33201							
READ	33204							
WRITE	33207							

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COPY 33212
 REM 33215
 WEF 33220
 BSR 33223
 BSF 33226
 STASH 33231
 COMENT 37522
 CLOCK 37521 03177 01663
 NONE 42720 00106 00070
 MINUTE 42720
 XMIN 42720

*LA10 * JUST LOADED.

UNUSED CORE LIES FROM 43026 THROUGH 50031, LEAVING 05004 OCTAL OR 02564 DECIMAL LOCATIONS.

ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY, REWINO SORT (DFSB) READS BSREC (DFAD) (DFMP) (DFDP) DRDD WRITB	ENDFIL	ENTRY NAME **PCB	ENTRY ADD. 22304	TRANSFER VECTORS EPOX TAPEX LOCO REDE REWIND KOOL DNTR2 DNTR1 TYPRYT EPPHEM FLAK SORT SORT GIG2 RCAL RCOM RSIG ACE	LOAD ADD. 22300	OCTAL LENGTH 00022	DECIMAL LENGTH 00018	COMMON BREAK
THE NAME OF THIS PROGRAM IS *LA11 * 2/22/65								
		LOGO	22332		22322	00165	00117	
		TRADE	22455					
		EPOX	22466					
		ACE	22514		22507	00067	00055	47675
		DMATRX	22610		22576	01432	00794	47675
		ERROR	24230	*NONE*	24230	00012	00010	
		ADD	24242	*NONE*	24242	00050	00040	
		ARSIN	24322	*NONE*	24312	00144	00100	77152
		ARCOS	24316					
		QARSIN	24314					
		QARCOS	24312					
		ERRARS	24443					
		ERRARC	24447					
		ARTAN	24456	*NONE*	24456	00103	00067	77152
		DAYS	24564	FIX	24561	00042	00034	
		EPPHEM	24644	FLOAT				
				ADD				
				QUIZ	24623	00604	00388	47675
				PONT				
				KINE				
				STARP				
				(DFSB)				
				TIMER				
				LOOKUP				
				INTR1				
				KOOL				
				GAMAT				
				RITEM				
				TDCIM				
				WREDF				
		FIXT	25430	FIXTT	25427	00007	00007	
		FIXTT	25436	*NONE*	25436	00312	00202	50031
		FLOT	25544					
		FIX	25725					
		FLOAT	25731					
		DAYE	25741					
		FLAT	25752	ERRDR	25750	00102	00066	
		FLAPR	26000	TYPRYT				
		FLAK	26007					
		GIG2	26056	*NONE*	26052	00266	00182	50001
		EPHEM	26344	REWIND	26340	01452	00810	
		TAPEX	27524	READB				
		PHUT	30020	BSREC				
		ROT	30354	SIN	30012	00336	00222	50001
		GAMAT	30652	COS	30350	00266	00182	50031
				(DFAD)	30636	00614	00396	47675
				(DFSB)				
				LOOKUP				
				INTR1				
				INSPC				
				DMATRX				
				(DFMP)				
				(DFDP)				
				(DFDP)	31452	00320	00208	50031
				(DFSB)				
				(DFMP)				
				(DFAD)				
				DRDD				
				COS				
		INSPC	31776	*NONE*	31772	00310	00200	50031
		DNTR2	36365	EPHEM	32302	04214	02188	
		INTR1	36406	PHUT				
				ROT				
				ERROR				
				REDE	36516	00200	00128	50001
				ERROR				
				ERROR	36716	00463	00307	47777
				REDE				
				DFESYS	37401	00311	00201	51235
				(DFAD)				
				TIMER				
				LOOKUP				
				INTR1				
				GAMAT				
				RITEM				

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QUIZ	37712	*NONE*	37712	00003	00003	
RCAL	37722	SQRT	37715	00240	00160	50031
RCOM	40162	SQRT	40155	00133	00091	50031
WREOF	40431	FLAT	40310	02520	01360	
RITEH	40314	WRITEB ENDFIL REHIND				
RSIG	43034	*NONE*	43030	00064	00052	47675
OFFSYS	43114	*NONE*	43114	00010	00008	
ENDSYS	43115					
FINSYS	43116					
SCHAHN	43117					
PGSTRY	43120					
PGSTOP	43121					
DCP	43122					
TYPRYT	43123					
SIN	43124	*NONE*	43124	00242	00162	
COS	43127					
QSIN	43133					
QCOS	43135					
STARP	43366	*NONE*	43366	00031	00025	
TIMER	43426	(DFSB)	43417	00142	00098	50031
		LOOKUP				
		SQRT				
TOCIM	43600	OFFSYS	43561	01126	00598	
		TIMER				
		LOOKUP				
		INTR1				
		SQRT				
		KOOL				
		CANAT				
		CHADP				
		COS				
		SIN				
		ARSIN				
		ARCOS				
		WRITEB				
		FLAT				
		RITE				
REDE	44710	DCP	44707	00421	00273	
RITE	44713					
PI5A	45137					
(DFAD)	45330	*NONE*	45330	00120	00080	77776
(DFSB)	45350					
(DFMP)	45370					
(DFDP)	45416					
DMDD	45450	*NONE*	45450	00060	00048	77776
SQRT	45530	*NONE*	45530	00055	00045	77773
READD	45606	(IDU)	45605	00365	00245	
READB	45610					
WRITED	45613					
WRITEB	45615					
BSREC	46024					
BSFILE	46027					
REHIND	46032					
UNLOAD	46035					
ENDFIL	46040					
SETLOW	46016					
SETHI	46021					
(UWIT)	46167					
TAPEID	46167					
(IDU)	46175	*NONE*	46172	00030	00024	

'LA11 ' JUST LOADED.

UNUSED CORE LIES FROM 46222 THROUGH 47675, LEAVING 01454 OCTAL OR 00812 DECIMAL LOCATIONS.

ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY.	DEFINE	ATTACH	OPEN	CLOSE	SQRT	PROUT	(DFSB)	(DFAD)	FGOUT	EXIT
THE NAME OF THIS PROGRAM IS 'LA12 ' 2/22/65										
ENTRY NAME	ENTRY ADD.	TRANSFER	VECTORS	LOAD ADD.	OCTAL LENGTH	DECIMAL LENGTH	COMMON BREAK			
*PCB	22302	SAVCOM		22301	00123	00083				
SAVCOM	22440	DEFINE		22424	02154	01132				
		ATTACH								
		OPEN								
		PRIM								
		UWFST								
		MAXIM								
		OFFSYS								
		TYPRYT								
		RITE								
		CLOSE								
		ERROR								
		FLAK								
BIRCD	24025	*NONE*		24600	00055	00045				
DIAGO	24662	SQRT		24655	00105	00069		50031		
ERROR	24766	TYPRYT		24762	00544	00356				
		PROJT								
		ENDSYS								
FLAT	25530	RECOV		25526	00102	00066				
FLAPR	25556	ERROR								
FLAK	25565	TYPRYT								
KTIME	25632									
MAXIM	26056	REDE		25630	00215	00141		50001		
		ERROR								
		QUIZ		26045	00473	00315		50001		
		XINE								
		(DFSB)								
		(DFAD)								
MOCT	26544	MOCT								
NORMAY	26654	*NONE*		26540	00107	00071				
APRED	27204	SORT		26647	00252	00170		77461		
UPREJ	27231	FIXT		27121	00143	00099				
OPERA	27122									
DUKEY	27246	*NONE*		27264	00040	00032				
OPKEY	27270									
COKEY	27305									
AFKEY	27312									
PRIM	27345	FGOUT		27324	01424	00788		50001		
PRAM	27334	PROUT								
		BIRCD								
		FLAPR								
		PRIME								
		NORMAY								
		PRIT								
		DIAGO								
TAMP	30751	*NONE*		30750	00004	00004				
PRIME	30752									
TAPR	30753									
DIAGO	*****									

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PRIT	30756	FIXT	30754	00035	00029	
QUEST	31020	BIFCO	31031	00165	00117	47675
		ERRRQ				
		EXIT				
QUIZ	31176	*NONE*	31176	00002	00002	
RECOV	31200	*NONE*	31200	00002	00002	
SETUP	31204	RITE	31202	00105	00069	50031
		ENDSYS				
FIXT	31310	FIXTT	31307	00007	00007	
FIXTT	31316	*NONE*	31316	00312	00202	50031
FLOT	31424					
FIX	31605					
FLOAT	31611					
DAYE	31621					
OFFSYS	31630	*NONE*	31630	00011	00009	
ENDSYS	31631					
FINSYS	31632					
SCHAIN	31633					
PGSTRT	31634					
PGSTOP	31635					
DCP	31636					
TYPRYT	31637					
REDE	31642	DCP	31641	00421	00273	
RITE	31645					
PISA	32071					
EXIT	32263	*NONE*	32262	00210	00136	
EKSEL	32366					
EXITFN	32274					
ERRRMP	32262					
(DFAD)	32472	*NONE*	32472	00120	00080	77776
(DFSB)	32512					
(DFMP)	32532					
(DFDP)	32560					
SQRT	32612	*NONE*	32612	00054	00044	77773
PROUT	32702	DUTUS	32666	03032	01562	
FGDOUT	34146	ACTIND				
PRCONV	33047	BFLC				
PRDUT2	33020	RESTKA				
PROUT3	33033	REGIND				
TSXA	35055	WRITE				
TSXB	35645	PACON				
		RGGSAV				
		RGGSTR				
		CKIND				
		CKACT				
DUTUS	35722	RGGSAV	35720	02320	01232	
BFLC	36954	RGGSTR				
ENDOUT	36307					
CKIND	40151					
CKACT	40214					
REGIND	40150					
ACTIND	40147					
RESTKA	40145					
PLICON	40230					
PL2CON	40231					
PL3CON	40232					
PRCON	40233					
RGGSAV	40240	*NONE*	40240	00133	00091	
RGGSTR	40322					
IOCS	40374	CLOCK	40373	04335	02269	
DEFINE	40375					
JOIN	40400					
ATTACH	40403					
CLOSE	40406					
OPEN	40411					
READ	40414					
WRITE	40417					
COPY	40422					
REW	40525					
WEP	40430					
BSR	40433					
BSF	40436					
STASH	40441					
CLOCK	44733	*NONE*	44730	00106	00070	
MINUTE	44730					
XMJH	44730					

*LA12 * JUST LOADED.

UNUSED CORE LIES FROM 45036 THROUGH 47675, LEAVING 02640 OCTAL OR 01440 DECIMAL LOCATIONS.

ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY.

DEFINE	ATTACH	OPEN	PROUT	CLOSE	(DFDP)	(DFSB)	(DFMP)	(DFAD)	DNOD
THE NAME OF THIS PROGRAM IS *ODPX * 2/22/65									
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.		OCTAL LENGTH	DECIMAL LENGTH		COMMON BREAK	
FIRST	22312	DATA	22301		02070	01080			
		DEFINE							
		ATTACH							
		OPEN							
		PROUT							
		NOHML							
		RITE							
		CLOSE							
		FLAX							
**PCB	24372	FIRST	24371		00007	00007			
ERROR	24400	*NONE*	24400		00012	00010			
FDATA	24412	*NONE*	24412		01117	00591			
FIXT	25532	FIXTT	25531		00007	00007			
FIXTT	25540	*NONE*	25540		00312	00202			50031
FLOT	25646								
FIX	26027								
FLOAT	26039								
DAYE	26043								
GHADP	26064	(DFDP)	26052		00320	00208			50031
		(DFSB)							
		(DFMP)							
		(DFAD)							
		DNOD							
		CDS							
NOHML	26376	*NONE*	26372		00763	00499			47777
NORWAY	27362	SQRT	27355		00251	00169			77461
QUIZ	27626	*NONE*	27626		00002	00002			
OFFSYS	27630	*NONE*	27630		00010	00008			
ENDSYS	27631								
FINSYS	27632								

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SCHAIN	27633				
PGSTRT	27634				
PGSTOP	27635				
QCP	27636				
TYPRYT	27637				
SIN	27640	'NONE'	27640	00243	00163
COS	27643				
QSIN	27647				
QCOS	27651				
REDS	30104	QCP	30103	00421	00273
RITE	30107				
PISA	30333				
FLAT	30526	ERROR	30524	00102	00966
FLAPR	30554	TYPRYT			
FLAK	30563				
IDFAD1	30626	'NONE'	30626	00120	00080
IDFS1	30646				77776
IDFMP1	30666				
IDFDP1	30714				
QRBO	30746	'NONE'	30746	00060	00948
QRBT	31026	'NONE'	31026	00054	00044
PRDUT	31116	DOTUS	31102	03032	01562
FDDUT	32262	ACTIND			
PRCNV	31263	BFLC			
PRDUT2	31234	RESTKA			
PRDUT3	31247	REQIND			
TSXA	33271	WRITE			
TSXB	34061	PRCON			
		RGGSAY			
		RUGSTR			
		CKIND			
		CKACT			
		RGGSAY	34134	02320	01232
		RUGSTR			
DOTUS	34136				
BFLC	34772				
ENDOUT	34523				
CKIND	36365				
CKACT	36430				
REQIND	36364				
ACTIND	36363				
RESTKA	36361				
PL1CON	36444				
PL2CON	36445				
PL3CON	36446				
PRCON	36447				
RGGSAY	36454	'NONE'	36454	00133	00091
RUGSTR	36536				
IGCS	36610	CLOCK	36607	04335	02269
DEFINE	36611				
JOIN	36614				
ATTACH	36617				
CLOSE	36622				
OPEN	36625				
READ	36630				
WRITE	36633				
COPY	36636				
REW	36641				
WFF	36644				
BSR	36647				
BSF	36652				
STASH	36659				
CLOCK	43147	'NONE'	43144	00106	00070
MINUTE	43144				
XMIN	43144				

*DDPX * JUST LOADED.

UNUSED CORE LINES FROM 43252 THROUGH 47777. LEAVING 04526 OCTAL OR 02390 DECIMAL LOCATIONS.

QCP COMMON MAP, JANUARY 4, 1965.

77461	I	COLUMN	2	RUNNING TIME
77457	EPOCH	COLUMN	2	EPOCH
77455	FINAL	COLUMN	2	FINAL TRAJECTORY TIME
77453	OBSERVATION	COLUMN	2	OBSERVATION TIME
77451	DELTA	COLUMN	2	LIGHT TIME CORRECTION
77447	LOWER	COLUMN	2	LOWER LIMIT OF LINE INTEGRATION
77445	DELTA	COLUMN	2	SECOND LIGHT TIME CORRECTION
				FILE TRANS CARD 02
77443	GHA	COLUMN	1	GREENWICH HOUR ANGLE
77442	GHA	COLUMN	1	INJECTION GHA
77441	RUTMX	COLUMN	3+1	PRED. OF PRECESSION AND NUTATION MX.
77430	ORIG	COLUMN	36	
77430	ORIG	COLUMN	36	
77427	ORIG	COLUMN	36	NOT ZERO FOR GEOCENTRIC INTEGRATION
77426	ORIG	COLUMN	36	NOT ZERO TO UPDATE CURRENT
77425	ORIG	COLUMN	36	BCD TARGET NAME
77424	ORIG	COLUMN	36	TRANSPONDER OFFSET
77423	ORIG	COLUMN	36	
77422	ORIG	COLUMN	36	TARGET (1=HARS 2=MLW 3=VENUS)
77421	ORIG	COLUMN	36	PHYSICAL CONSTANTS STEP MULTIPLIER
77420	ORIG	COLUMN	36	NOT ZERO FOR FLOATING RESIDUALS
77417	ORIG	COLUMN	36	NOT ZERO INDICATES OPTICAL DATA
77416	ORIG	COLUMN	36	NOT ZERO TO PUNCH J MATRIX
77415	ORIG	COLUMN	36	
77414	ORIG	COLUMN	36	
77413	ORIG	COLUMN	36	INITIAL TRAJ STEP SIZE
77412	ORIG	COLUMN	36	NOT ZERO FOR MMP OUTPUT
77411	ORIG	COLUMN	36	UP LEG SLANT RANGE
77410	ORIG	COLUMN	36	DOWN LEG SLANT RANGE
77407	ORIG	COLUMN	36	UP LEG REFRACTION CORRECTION
77406	ORIG	COLUMN	36	DOWN LEG REFRACTION CORRECTION
77405	ORIG	COLUMN	36	TELE-EPOLM
77404	ORIG	COLUMN	36	UP
77403	ORIG	COLUMN	36	NOT FROM SPACE
77402	ORIG	COLUMN	36	DOWN FROM SPACE
77401	ORIG	COLUMN	36	
77400	ORIG	COLUMN	36	
77377	ORIG	COLUMN	36	
77376	ORIG	COLUMN	36	
77375	ORIG	COLUMN	36	NOT ZERO, OCCULTATION-IMPACT DATA
77374	ORIG	COLUMN	36	
77373	ORIG	COLUMN	36	
77372	ORIG	COLUMN	36	
77371	ORIG	COLUMN	36	HARDWARE NUMBER STARTER
77370	ORIG	COLUMN	36	SUBR. ERROR 1+4
77367	ORIG	COLUMN	36	SUBR. ERROR 2+4
77366	ORIG	COLUMN	36	SUBR. ERROR 3+4
77365	ORIG	COLUMN	36	EPOCH YTHODDHH
				EPOCH MMSSTFF
				FUNCTION CARD 03
77364	SLANT	COLUMN	1	SLANT RANGE
77363	SLANT	COLUMN	1	SLANT RANGE RATE
77362	ELEVATION	COLUMN	1	ELEVATION ANGLE
77361	AZIMUTH	COLUMN	1	AZIMUTH ANGLE

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77360	DEC	COMMON	1	LOCAL DECLINATION ANGLE
77357	HA	COMMON	1	LOCAL HOUR ANGLE
77356	C1	COMMON	1	1 WAY DOPPLER
77355	CC3	COMMON	1	COHERENT 3 WAY DOPPLER
77354	C3	COMMON	1	3 WAY DOPPLER
77353	CO1	COMMON	1	DIFFERENCED 1 WAY DOPPLER
77352	CO3	COMMON	1	DSIF RANGING
77351	XXX	COMMON	1	EXTRA DATA TYPE
* FORTRAN CARD 04				
77350	DRVEC	COMMON	1	CORRECTIONS
77347	DRDDT	COMMON	1	
77346	DEL	COMMON	1	
77345	DAZ	COMMON	1	
77344	DDEC	COMMON	1	
77343	DHA	COMMON	1	
77342	DC1	COMMON	1	
77341	DC3	COMMON	1	
77340	DC3	COMMON	1	
77337	DC01	COMMON	1	
77336	DC03	COMMON	1	
77335	DXXX	COMMON	1	
* FORTRAN CARD 05				
77334	RVEC0	COMMON	1	OBSERVED VALUES
77333	RDDTU	COMMON	1	
77332	ELU	COMMON	1	
77331	AZIU	COMMON	1	
77330	DECU	COMMON	1	
77327	HAD	COMMON	1	
77326	C1U	COMMON	1	
77325	CC3U	COMMON	1	
77324	C3U	COMMON	1	
77323	CO1U	COMMON	1	
77322	CO3U	COMMON	1	
77321	XXU	COMMON	1	
* FORTRAN CARD 06				
77320	RVEC0	COMMON	1	RESIDUALS (OBSERVED-CALCULATED)
77317	RDDTD	COMMON	1	
77316	ELU	COMMON	1	
77315	AZIU	COMMON	1	
77314	DECU	COMMON	1	
77313	HAD	COMMON	1	
77312	C1U	COMMON	1	
77311	CC3U	COMMON	1	
77310	C3U	COMMON	1	
77307	CO1U	COMMON	1	
77306	CO3U	COMMON	1	
77305	XXU	COMMON	1	
* FORTRAN CARD 07				
77304	RVEC0	COMMON	1	WEIGHTS
77303	RDDTD	COMMON	1	
77302	ELU	COMMON	1	
77301	AZIU	COMMON	1	
77300	DECU	COMMON	1	
77277	HAD	COMMON	1	
77276	C1U	COMMON	1	
77275	CC3U	COMMON	1	
77274	C3U	COMMON	1	
77273	CO1U	COMMON	1	
77272	CO3U	COMMON	1	
77271	XXU	COMMON	1	
* FORTRAN CARD 08				
77270	IX	COMMON	1	RECEIVER I.O.
77267	IX	COMMON	1	EMITTER I.O.
77266	ID	COMMON	1	CURRENT DATA TYPE
77265	PASS	COMMON	1	PASS IDENTIFICATION
77246	XJUP	COMMON	6	JUPITER VECTOR
77240	XERS	COMMON	6	SATURN VECTOR
77232	XMAR	COMMON	6	MARS VECTOR
77224	XVEN	COMMON	6	VENUS VECTOR
77216	XSN	COMMON	6	SUN VECTOR
77210	XMO	COMMON	6	MOON VECTOR
* FORTRAN CARD 09				
77202	X	COMMON	1	PROBE VECTOR
77201	Y	COMMON	1	
77200	Z	COMMON	1	
77177	XDUT	COMMON	1	
77176	YDUT	COMMON	1	
77175	ZDUT	COMMON	1	
77174	U	COMMON	36	VARIATIONAL EQUATIONS
77130	DL0	COMMON	1	NUTATION IN LONGITUDE
77127	DL8	COMMON	1	NUTATION IN OBLIQUITY
* FORTRAN CARD 10				
77126	XAC	COMMON	1	PROBE ACCELERATION
77125	YAC	COMMON	1	
77124	ZAC	COMMON	1	
77123	XJERK	COMMON	1	PROBE JERK (THIRD DERIVATIVE)
77122	YJERK	COMMON	1	
77121	ZJERK	COMMON	1	
77120	UINV	COMMON	36	INVERSE OF VARIATIONAL EQUATIONS
77054	PUBX	COMMON	6*12	PARTIALS(OBSERVABLES/CARTESIAN)
76744	BMAT	COMMON	6*11	B MATRIX OF PARTIALS
* FORTRAN CARD 11				
76642	ELX	COMMON	1	DIRECTION COSINES
76641	ELY	COMMON	1	
76640	ELZ	COMMON	1	
76637	AX	COMMON	1	DIRECTION COSINES
76636	AY	COMMON	1	
76635	AZ	COMMON	1	
76634	DX	COMMON	1	DIRECTION COSINES
76633	DY	COMMON	1	
76632	DZ	COMMON	1	
76631	TAX	COMMON	1	DIRECTION COSTINES
76630	TAY	COMMON	1	
76627	TAZ	COMMON	1	
76626	TDX	COMMON	1	DIRECTION COSINES
76625	TDY	COMMON	1	
76624	TDZ	COMMON	1	
* FORTRAN CARD 12				
76623	TAU	COMMON	1	DOPPLER AVERAGING TIME
76622	LLIST	COMMON	6	POSITION AND VELOCITY FLAGS
76614	MLIST	COMMON	11	PHYSICAL CONSTANTS PARTIALS FLAGS
76601	HLIST	COMMON	1	VELOCITY OF LIGHT FLAG

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76600	MLIST	COMMON	3=15	STATION LOCATION FLAGS
76523	NEST	COMMON	1	NO. ESTIMATED PARAMETERS
76522	NCUM	COMMON	1	NO. CONSIDERED PARAMETERS
76521	EP	COMMON	100	E.T.-U.T.
76520	EGU		EP-1	
76513	EGU		EP-6	BAND(-1017=LS 0=L 1017=S)
76512	EGU		EP-7	K1 FREQUENCY
76511	EGU		EP-4	K5 FREQUENCY
76510	EGU		EP-9	BAND DESIGNATION (15 WORDS)
76471	EGU		EP-24	START TIME, ATTITUDE CONTROL
76470	EGU		EP-25	UP
76467	EGU		EP-26	STOP TIME, ATTITUDE CONTROL
76466	EGU		EP-27	DP
76465	EGU		EP-28	
76457	EGU		EP-34	MCIM COMMENT BUFFER (30 WORDS)
76421	EGU		EP-64	NOT ZERO TO READ CARDS FROM A2
76420	EGU		EP-65	ID FOR PREDICTS
76417	EGU		EP-66	POWERED FLIGHT PARAMETERS (10 WORDS)
76375	ELU		EP-84	STEP MAP FLAG=U PRODUCE
76374	EGU		EP-95	START TIME, STEP MAP PRINTOUT
76373	EGU		EP-96	DP
76372	EGU		EP-97	FLAG TO REPEAT TRAJ FOR STEP MAP
76371	ELU		EP-98	SPACE COMMENT BUFFER (17 WORDS)
76350	ELU		EP-105	MAPPLD FORWARD EPU=H
76347	EGU		EP-106	DP
76346	EGU		EP-107	FLAG FOR PUNCHING MATRICES
76345	EGU		EP-108	NOT ZERO TO BYPASS LALL
76344	EGU		EP-109	
76235	SIGM3	COMMON	12=15	REJECTION SIGMAS
				FORTRAN CARD 13.
75751	A	COMMON	1	COMMON
75750	B	COMMON	1	STORAGE
75747	C	COMMON	1	AVAILABLE
75746	D	COMMON	1	FOR
75745	E	COMMON	1	FAP
75744	F	COMMON	1	OR
75743	G	COMMON	1	FORTRAN
75742	H	COMMON	1	USAGE
75741	D	COMMON	1	
75740	P	COMMON	1	
75737	Q	COMMON	1	
75736	R	COMMON	1	
75735	S	COMMON	1	
				FORTRAN CARD 14
75734	RNJ	COMMON	2	SPHERICAL INJECTION CONDITIONS
75733	PHNJ	COMMON	2	
75730	IHNJ	COMMON	2	
75726	VNJ	COMMON	2	
75724	ELNJ	COMMON	2	
75722	AZNJ	COMMON	2	
75720	XNJ	COMMON	6	GAUSSIAN INJECTION CONDITIONS
75712	XNDP	COMMON	6	
				FORTRAN CARD 15
75704	GRAV	COMMON	1	K(EARTH)
75703	RE	COMMON	1	EARTH RADIUS
75702	GMB	COMMON	1	RADIATION PRESSURE CONSTANT
75701	GRAVP	COMMON	1	K(MOON)
75700	KVEN	COMMON	1	MASS RATIO VENUS
75677	KMAR	COMMON	1	MASS RATIO MARS
75676	KJUP	COMMON	1	MASS RATIO JUPITER
75675	UJ	COMMON	1	Coefficient SECOND HARMONIC
75674	UH	COMMON	1	Coefficient THIRD HARMONIC
75673	UD	COMMON	1	Coefficient FOURTH HARMONIC
75672	AU	COMMON	1	ASTRONOMICAL UNIT
75671	VELC	COMMON	1	VELOCITY OF LIGHT
				FORTRAN CARD 16
75670	RI	COMMON	15	STATION RANGE
75651	PHI1	COMMON	15	STATION LATITUDE (GEOCENTRIC)
75632	THETA1	COMMON	15	STATION LONGITUDE
				FORTRAN CARD 17
75613	GRAVS	COMMON	1	K(SUN)
75612	GRAVV	COMMON	1	K(venus)
75611	GRAVW	COMMON	1	K(MARS)
75610	GRAVJ	COMMON	1	K(JUPITER)
75607	OMEGA	COMMON	1	ANGULAR ROTATION RATE OF EARTH
75606	XI	COMMON	1	STATION X(1)
75605	YI	COMMON	1	STATION Y(1)
75604	ZI	COMMON	1	STATION Z(1)
75603	RAI	COMMON	1	STATION RIGHT ASCENSION
				FORTRAN CARD 18
75602	UI	COMMON	15	NORTH SOUTH DISPLACEMENT (GEO-GEOC)
75563	VI	COMMON	15	EAST WEST DISPLACEMENT
75544	FNI	COMMON	15	LOCAL INDEX OF REFRACTION
75525	TAB	COMMON	1000	
75525	EGU		TAB	DATA FILE INPUT BUFFER (400 WORDS)
74705	EGU		TAB=400	DATA FILE OUTPUT BUFFER (400 WORDS)
74065	EGU		TAB=800	SCRATCH AREA, LAB (200 WORDS)
73555	YI	COMMON	4=24	PROBE EPHEMERIS INTERPOLATION BUFFER
71325	RESK	COMMON	8=15	SCALE FACTORS
71135	IRES	COMMON	8=15	PLOT FLAGS
70745	NHR	COMMON	1	PLOT STEP
				FORTRAN CARD 19
70744	DRBL	COMMON	12	VELOCITY OF LIGHT PARTIALS
70730	DIM3	COMMON	12=3*15	STATION PARTIALS BY DATA TYPE
67674	CPPH1	COMMON	20	CAP PHI MATRIX OF PARTIALS
67650	CPPH2	COMMON	20	UP
67624	CPHI1	COMMON	20	CAP THETA MATRIX OF PARTIALS
67600	CPHI2	COMMON	20	DP
67554	XJ	COMMON	20=20	J MATRIX
66734	XJ2	COMMON	20=20	DP
66114	XK	COMMON	400	SCRATCH 20*20 MATRIX
65274	XK2	COMMON	400	SCRATCH 20*20 MATRIX
64454	RIGH1	COMMON	20	RESIDUAL=CPPH1 ACCUMULATED
64430	RITE2	COMMON	20	DP
				FORTRAN CARD 20
64404	CPGAM	COMMON	20=20	COVARIANCE MATRIX
63564	CPGM2	COMMON	20=20	UP
62744	GFLIP	COMMON	400	INPUT COVARIANCE MATRIX (11=VERSE)
62124	GFLIP?	COMMON	400	DP
61304	GAM1	COMMON	20=20	COV PX OF UNADJUSTED PARAMETERS
60464	DIAT	COMMON	60	V MATRIX 1=TEGARD

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60362	DIN2	COMMON	66	DP
60260	DIG	COMMON	48	
60260	EDU		DIG	
60231	PLOFIL	EQU	DIG-23	IOCS PLOT CONTROL (12 WORDS FORWARD)
60215	PUNFIL	EQU	DIG-35	IOCS PUNCH CONTROL (12 WORDS FORWARD)
60201	PRIFIL	EQU	DIG-47	IOCS PRINT CONTROL (12 WORDS FORWARD)
			FORTRAN CARD 21	
60200	YEAR	COMMON	1	7-WORD TIME BLOCK
60177	MONTH	COMMON	1	INTEGERS IN ADDRESS PART
60176	DAY	COMMON	1	FAP USE ONLY
60175	HOUR	COMMON	1	
60174	MIN	COMMON	1	
60173	SEC	COMMON	1	
60172	FRAC	COMMON	1	
			FORTRAN CARD 22	
60171	IF1	COMMON	1	
60170	IF2	COMMON	1	
60167	IF3	COMMON	1	CONTENTS OF KEYS 10-17
60166	IF4	COMMON	1	
60165	IF5	COMMON	1	ZERO TO CLOSE DATA FILE/PREDICT OUTPUT
60164	IF6	COMMON	1	DATA FILE READ 1617=EOF 2017=NORMAL
60163	IF7	COMMON	1	
60162	IF8	COMMON	1	NOT ZERO TO CHECK RESIDUAL REJECTION
60161	IF9	COMMON	1	
60160	IF10	COMMON	1	NOT ZERO TO PRINT PHI VECTORS
			FORTRAN CARD 23	
60157	KLIST	COMMON	4	FLAGS FOR ATTITUDE CONTROL ESTIMATION
60153	FABC	COMMON	9	ATTITUDE CONTROL COEFFICIENTS
60142	ELMIN	COMMON	1	MINIMUM ELEVATION ANGLE
60141	RMAX	COMMON	1	MAXIMUM SLANT RANGE
60140	RUN	COMMON	1	DATA FILE OUTPUT COUNTER
60137	BEFOR	COMMON	1	
60136	CA	COMMON	11	PARTIALS DF/DQ
60123	QB	COMMON	6	PARTIALS DF/DXO
60115	TOBL	COMMON	2	PREVIOUS TUB
60113	XMU	COMMON	20	SCRATCH COLUMN VECTOR
60067	XX	COMMON	400	SCRATCH 20*20 MATRIX
57247	XC	COMMON	400	SCRATCH 20*20 MATRIX
			FORTRAN CARD 24	
56427	IGNA2	COMMON	12*15	NOT ZERO TO IGNORE DATA POINT
56143	SOUR	COMMON	2	SUM OF SQUARED RESIDUALS
56141	BU225	COMMON	15*15	INPUT BUFFER
56000	BU12	COMMON	12	DP INPUT BUFFER
			FORTRAN CARD 25	
55564	XN	COMMON	3	CORRECTED PROBE COORDINATES
55561	DXN	COMMON	3	
55556	DOXN	COMMON	3	
55553	DDOXN	COMMON	3	
55550	IS	COMMON	1	DATA SAMPLE INTERVAL
55547	SSQ	COMMON	12*6*4	S=2 WEIGHTING TABLE
55107	TL	COMMON	12*6*4	T WEIGHTING TABLE
54447	GSQ	COMMON	6	G WEIGHTING TABLE
54441	OHBAO	COMMON	12	NOT ZERO IF DATA TYPE REJECTED LAST
			FORTRAN CARD 26	
54425	QTILD	COMMON	20	A PRIORI ESTIMATE
54401	QTP	COMMON	20	DP
54355	QP	COMMON	20	ESTIMATED CONDITIONS AND CONSTANTS
54331	QPDQ	COMMON	20	
54305	CZ1	COMMON	1	DOPPLER COEFFICIENTS
54304	CZ2	COMMON	1	
54303	CZ3	COMMON	1	
54302	CZ4	COMMON	1	
54301	CZ5	COMMON	1	
54300	CZ6	COMMON	1	
54277	PI	COMMON	45	INITIAL POINTING TIMES
54222	PID	COMMON	45	POINTING COUNT INTERVALS
54145	PIF	COMMON	45	FINAL POINTING TIMES
54070	PTFD	COMMON	45	
54070	EQU	PTFD		OCULTATION-IMPACT TIMES (30 WORDS)
54032	EQU	PTFD-30		START-STOP TIMES (12 WORDS)
54016	EQU	PTFD-42		
54013	PINT	COMMON	45	POINTING SAMPLE INTERVAL
53736	PINTU	COMMON	45	CURRENT POINTING TIME
53661	BALN	COMMON	12*15	INPUT WEIGHTS
			FORTRAN CARD 27	
53375	COSRA1	COMMON	1	COS(RA1)
53374	SINRA1	COMMON	1	SIN(RA1)
53373	COSPH1	COMMON	1	COS(PHI1)
53372	SINPH1	COMMON	1	SIN(PHI1)
53371	SINEL	COMMON	1	SIN(EL)
53370	CL37	COMMON	1	NOT ZERO TO BYPASS LAL3
53367	GADT	COMMON	1	PHYSICAL CONSTANTS INTEGRATION STEP
53366	CAPQ	COMMON	1	Q SUM OF SQUARES
53365	CAPQL	COMMON	1	
53364	DCQ	COMMON	1	
			FORTRAN CARD 28	
53363	NTDT	COMMON	1	NUMBER OF POINTS IN SOUR
53362	QNU	COMMON	20	NEW Q-S (QP+RIGHT)
53366	QNUD	COMMON	20	DP
53312	ITRU	COMMON	1	ITERATION NUMBER
53311	COSEL	COMMON	1	COS(EL)
53310	SINAZ	COMMON	1	SIN(AZ)
53307	COSAZ	COMMON	1	COS(AZ)
			FORTRAN CARD 29	
53306	CORR	COMMON	400	SCRATCH 20*20 MATRIX
52466	GTPP	COMMON	400	SCRATCH 20*20 MATRIX
51646	GMAI	COMMON	66	V MATRIX OF PARTIALS
51544	IFA	COMMON	10	ZERO TO INTEGRATE TO ENCOUNTER
51544	EQU	IFA		
51543	EQU	IFA-1		
51542	EQU	IFA-2		
51541	EQU	IFA-3		
51540	EQU	IFA-4		
51537	EQU	IFA-5		NOT ZERO FOR DATA SIGMA OUTPUT
51536	EQU	IFA-6		RADIUS OF TARGET
51535	EQU	IFA-7		NOT ZERO TO TURN OFF SOLAR PRESSURE
51534	EQU	IFA-8		NOT ZERO TO TURN OFF LIGHT TIME
51533	EQU	IFA-9		NOT ZERO TO TURN OFF REFRACTION
51532	XFREW	COMMON	1	XMITTER FREQUENCY
51531	YFREW	COMMON	1	YPOUNDER FREQUENCY
51530	KEYS	COMMON	1	CONTENTS OF KEYS
51527	XPBXX	COMMON	6	XMITTER ROOT PARTIALS
51521	YXCN	COMMON	6*15	FREQUENCY COEFFICIENTS
51367	YXCNZ	COMMON	6*15	DP
			FORTRAN CARD 30	
51235	COMTS	COMMON	11	PAGE HEADING
51222	SIGMA	COMMON	12*15	DATA FILE SIGMA
50736	BIAS	COMMON	12*15	DATA FILE BIAS
50452	IMAP	COMMON	2	CURRENT MAPPING TIME
50450	ITERAT	COMMON	1	
50447	YY1	COMMON	6*45	TRANSMITTER ID TABLE
50031	IMAP	COMMON	24	MAPPING TIMES
50001	AREA	COMMON	1	S/C AREA
50000	FMASS	COMMON	1	S/C MASS
47777	GAMM	COMMON	66	PHYSICAL CONSTANTS INTEGRATION MATRIX
47675	CAPU	COMMON	400	MAPPING MATRIX TO ENCOUNTER
			END	

V. OUTPUT

Primary output from the ODP is in the form of a BCD tape which may be listed on the IBM 1401. This listing contains the following items:

1. List of card input
2. A priori covariance matrix
3. Inverse of a priori covariance matrix
4. Trajectory information
5. J matrix
6. J matrix correlations
7. Parameter estimates and statistics
8. Inverse of J matrix
9. Covariance matrix
10. Covariance matrix correlations

The following items are output on option:

1. Residuals, computed observables, weights, and frequencies
2. Residual statistics
3. Pointing ephemeris
4. Mapping matrix
5. Mapped covariance matrix
6. Position and velocity of probe at mapping times
7. Encounter parameters and statistics
8. Supplementary printout at data times--vectors and partials

All items, with the exception of the supplementary partials printout, may be printed on the SC-3070 in SFOF Mode II operation.

Any matrix which is printed by the ODP may also be punched on option. In this case the card images are written on the same BCD output tape for punching by the 1401.

The ODP also writes a plotting tape for the SC-4020 plotter, containing plots of residuals versus time, and angle residuals versus angles.

A binary tape for the Matrix Manipulation Program, containing data partials and mapping matrices, is generated on option. This capability does not exist in Mode II operation.

Error comments and status-of-program messages are printed on the administrative printer in Mode II and the on-line printer in Mode IV. Refer to Subroutine 28, ERROR, for a listing of these comments.

A. DISK AND TAPE FORMATS

The Planetary Ephemeris format is presented in Ref. 4, and the Matrix Manipulation Program tape format in Ref. 11 (Section IX). The other formats are covered in Section VI; the Prediction File is presented in Subroutine 30, FILL; the Data File in Subroutine 43, KINE; the Probe Ephemeris in Subroutine 45, LOOKUP; and the Combined Ephemeris in Subroutine 86, RITEM.

B. OUTPUT SYMBOLS AND DEFINITIONS

The following symbols appear on the output listings. The corresponding symbols used in this report and the definitions are given:

Output symbol	Report symbol	Definition
X	x } y } z }	position of probe, geocentric equatorial true of date, km
Y		
Z		
DX	\dot{x} } \dot{y} } \dot{z} }	velocity of probe, geocentric equatorial true of date, km/sec
DY		
DZ		
KE	GM_e	gravitational constant of Earth, km^3/sec^2
RE	R_e	equatorial radius of Earth, km
G	γ_B	solar pressure constant
KM	GM_m	gravitational constant of Moon, km^3/sec^2
MV	M_v	mass of Venus, solar masses
MM	M_r	mass of Mars, solar masses

Output symbol	Report symbol	Definition		
MJ	M_j	mass of Jupiter, solar masses		
J	J	second harmonic coefficient		
H	H	third harmonic coefficient		
D	D	fourth harmonic coefficient		
AU	a_e	astronomical unit, km		
FA	F_{ao}	attitude		
FB			F_{bo}	control
FC				
C	c	speed of light, km/sec		
RI(I)	R_i	radius of Earth at station i, km		
LA(I)	ϕ_i	geocentric latitude of station i, deg		
LO(I)	λ_i	longitude of station i, deg		
R	ρ	slant range, km		
DR	$\dot{\rho}$	slant range rate, km/sec		
EL	γ	elevation angle, deg		
AZ	σ	azimuth angle, deg		
DEC	δ	declination, deg		
HA	α	hour angle, deg		
C1	f_1	one-way integrated doppler frequency, cps		
CC3	f_{c3}	coherent three-way integrated doppler frequency, cps		
C3	f_3	three-way integrated doppler frequency, cps		
D1	f_{d1}	differenced one-way integrated doppler frequency, cps		

Output symbol	Report symbol	Definition
RU	ρ DSIF	DSIF ranging
B	$ \underline{B} $	vector from target center of mass perpendicular to probe asymptote, km
B·RO	$\underline{B} \cdot \underline{R}_0$	dot products, target orbital plane, km
B·TO	$\underline{B} \cdot \underline{T}_0$	
B·RT	$\underline{B} \cdot \underline{R}_T$	dot products, target equatorial plane, km
B·TT	$\underline{B} \cdot \underline{T}_T$	
TL	t_L	linearized time of flight, hours (lunar missions) or days
TF	t_f	true time of flight, hours (lunar missions) or days
SMAA	a	semi-major axis of dispersion ellipse at target, km
SMIA	b	semi-minor axis, km
THETA	θ	inclination of dispersion ellipse to target orbital plane, deg
DEL T	σ_t	standard deviation of linearized time of flight, sec
DEL B	σ_B	standard deviation of \underline{B} vector, km
DEL S	σ_S	standard deviation of asymptote unit vector, km
DEL BR	$\sigma_{\underline{B} \cdot \underline{R}}$	standard deviation of $\underline{B} \cdot \underline{R}$, km
DEL BT	$\sigma_{\underline{B} \cdot \underline{T}}$	standard deviation of $\underline{B} \cdot \underline{T}$, km
C3	c_3	vis viva energy, km^2/sec^2
TC	τ	doppler count time, sec
Q	q	transmitter index
FRQ	$\left\{ \begin{array}{l} f_q \\ f_T \end{array} \right.$	if $q \neq 0$, station frequency less 29.66×10^6 cps
		if $q = 0$, probe frequency less 960×10^6 cps

VI. SUBROUTINES

Section VI contains the documentation for the 104 Subroutines which were written primarily for use by the SPODP. Documentation is not included for JPL general-purpose routines and SFOF routines.

IDENTIFICATION

1

AAT
Melba Nead, JPL
Fortran II, Version 3
January 4, 1965

PURPOSE

This subroutine forms the matrix product of two vectors.

RESTRICTIONS

Standard Fortran II arrays.

USE

CALL	AAT (X, Y, PRO)
X (1, 6)	P
Y (6, 1)	P ^T
PRO (6, 6)	Product matrix

CODING INFORMATION

Length of subroutine is 77 (10) or 115 (8) words.

IDENTIFICATION

2

ABDD

Charles Coltharp, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

ABDD computes the function.

$$S = \frac{Z_i(ax + by) - cz}{d},$$

the form of which has frequent application in ODP station location partial expressions.

RESTRICTIONS

COMMON break: 47675

USE

CALL ABDD

PZE D d

PZE R x, y, z (1 x 3)

COMMON input:

O a

B b

C c

Z_i Z - component of geocentric station vector, km.

COMMON output:

S s

CODING INFORMATION

Length of subroutine is 24 (10) or 30 (8) words.

IDENTIFICATION

3

ACE

Melba Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To compute the unit vector as needed in calculation of the attitude control effects partials in BMATRIX.

RESTRICTIONS

- a. COMMON from ODP
- b. Subroutines used:
 SQRT

USE

CALL ACE (DDX)

Computes R, X/R etc, and stores the result in DDX(I).

CODING INFORMATION

Length of subroutine is 57 (10) or 71 (8) words.

IDENTIFICATION

4

ANGPLT

Alfred Schoepke, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

ANGPLT plots hour angle residuals versus hour angle, and declination residuals versus declination. A tape containing the plotting information is written for the SC-4020.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutine used:

PKINE

FIXT

BIBCD

CXPLOT

USE

CALL ANGPLT

COMMON input:

IR station identification

RVECD residual array

RVEC data type array

CODING INFORMATION

Length of subroutine is 383 (10) or 577 (8) words.

IDENTIFICATION

5

AQUI/ALLI/BAQUI/BALLI

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

To format, list, and send, via TELTYP, predictions needed by the tracking stations.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

PROUT

FGDOUT

PKINE

FIXT

TELTYP

ENDOUT

BIBCD

In MODE 2, the predictions are transmitted in real time by TELTYP. In MODE 4, a magnetic tape (A9) is written by TELTYP. This tape is then processed to punch paper tape acceptable to the teleprinters.

USE

CALL AQUI to list predictions

CALL BAQUI to prepare for transmission

CALL ALLI to list drive tape for Goldstone

CALL BALLI to prepare drive tape for transmission

CODING INFORMATION

Length of subroutine is 460 (10) or 714 (8) words.

REFERENCE

Heller, J., IOM to F. Curl, Nominal Predict Format, June 10, 1963.

IDENTIFICATION

6

BEER

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

BEER stores zeroes in certain ODP common locations prior to the data fit. It also executes the reset entry of REJEC.

RESTRICTIONS

- a. COMMON break: 46711
- b. Subroutine called:
REJEC

USE

CALL BEER

COMMON output: zeroes in DINT, DIN2, QA, QB, TOBL, TOBL-1, TGINT, TGINT-1, XMU, RIGHT, RITE2, XX, XC, XJ, XJ2, XK, XK2, IF2, IF4. BIGNO in IF1 and IF5.

CODING INFORMATION

Length of subroutines is 57 (10) or 71 (8) words.

IDENTIFICATION

7

BIBCD

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

To convert the binary equivalent of 2 digit decimal numbers to BCD for print-out.

RESTRICTION

Range of numbers: $1 \leq n \leq 99$

USE

With argument in the accumulator

CALL BIBCD

The BCD equivalent is in the accumulator on return.

CODING INFORMATION

Length of subroutine is 45 (10) or 55 (8) words.

IDENTIFICATION

8-1 of 4

BMATRIX

Melba Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

BMATRIX is written to compute the analytic partials of the physical constants when these parameters are being estimated or considered in the ODP.

RESTRICTIONS

- a. COMMON break: 47675
- b. When computing partials of the attitude control effects, partials of MJ, H, and D are inhibited, thus a maximum of 11 of the possible 14 can be computed at one time.
- c. Subroutines used:

ACE
GIG2
RCAL
RCOM
RSIG
SQRT

METHOD

Formulas for the partial derivatives of acceleration with respect to physical constants, $\partial \ddot{r}/\partial q$, are derived from the equations of motion. These are integrated to obtain $\partial r/\partial q$.

The equations for these partials follow:

$$\frac{\partial \ddot{x}}{\partial GM_e} = \frac{-xg_1}{r^3}$$

$$\frac{\partial \ddot{y}}{\partial GM_e} = \frac{-yg_1}{r^3}$$

$$\frac{\partial \ddot{z}}{\partial GM_e} = \frac{-zg_2}{r^3}$$

$$\frac{\partial \ddot{x}}{\partial R_e} = \frac{-GM_m}{R_e} \left\{ 2x_m \left[\frac{1}{|r_m - r|^3} - \frac{1}{r_m^3} \right] - 3 \frac{(r \cdot r_m - r^2)(x - x_m)}{|r_m - r|^5} - \frac{3x}{|r_m - r|^3} \right\} \quad x \rightarrow y, z$$

8-2 of 4

$$\frac{\partial \ddot{x}}{\partial \gamma_B} = \frac{c_1 A_p (x - x_s)}{M_p |r - r_s|^3}$$

x → y, z

where

$$c_1 = 1.031 \times 10^8$$

A_p = surface area of probe, m^2

M_p = mass of probe, kg

$$\frac{\partial \ddot{x}}{\partial GM_m} = - \left[\frac{x_m}{r_m^3} + \frac{(x - x_m)}{|r_m - r|^3} \right]$$

x → y, z

$$\frac{\partial \ddot{x}}{\partial M_v} = -GM_s \left[\frac{x_v}{r_v^3} + \frac{(x - x_v)}{|r_v - r|^3} \right]$$

x → y, z

$$\frac{\partial \ddot{x}}{\partial M_r} = -GM_s \left[\frac{x_r}{r_r^3} + \frac{(x - x_r)}{|r_r - r|^3} \right]$$

x → y, z

$$\frac{\partial \ddot{x}}{\partial M_j} = -GM_s \left[\frac{x_j}{r_j^3} + \frac{(x - x_j)}{|r_j - r|^3} \right]$$

x → y, z

$$\frac{\partial \ddot{x}}{\partial J} = -x \left[1 - 5 \left(\frac{z}{r} \right)^2 \right] \frac{GM_e R_e^2}{r^2 r^3}$$

x → y

$$\frac{\partial \ddot{z}}{\partial J} = -z \left[3 - 5 \left(\frac{z}{r} \right)^2 \right] \frac{GM_e R_e^2}{r^2 r^3}$$

$$\frac{\partial \ddot{x}}{\partial H} = x \left[7 \left(\frac{z}{r} \right)^2 - 3 \right] \frac{z}{r} \frac{GM_e R_e^3}{r^3 r^3}$$

x → y

$$\frac{\partial \ddot{z}}{\partial H} = \left[1 - 10 \left(\frac{z}{r} \right)^2 + \frac{35}{3} \left(\frac{z}{r} \right)^4 \right] \frac{3z}{5} \frac{GM_e R_e^3}{r^3 r^3}$$

$$\frac{\partial \ddot{x}}{\partial D} = -x \left[3 - 42 \left(\frac{z}{r} \right)^2 + 63 \left(\frac{z}{r} \right)^4 \right] \frac{GM_e R_e^4}{7r^3 r^4}$$

x → y

$$\frac{\partial \ddot{z}}{\partial D} = -z \left[15 - 70 \left(\frac{z}{r} \right)^2 + 63 \left(\frac{z}{r} \right)^4 \right] \frac{GM_e R_e^4}{7r^3 r^4}$$

$$\frac{\partial \ddot{x}}{\partial a_e} = \frac{1}{a_e} \sum_{k=1}^4 GM_k \left[x_k \left(\frac{1}{|r_k - r|^3} - \frac{1}{r_k^3} \right) - \frac{3(r^2 - r \cdot r_k)(x - x_k)}{|r_k - r|^5} \right] \quad x \rightarrow y, z$$

where

- k = 1 = Sun
- k = 2 = Venus
- k = 3 = Mars
- k = 4 = Jupiter

The partial derivatives of acceleration with respect to the attitude control effects are given by

$$\begin{bmatrix} \frac{\partial \ddot{x}}{\partial F_{AO}} & \frac{\partial \ddot{x}}{\partial F_{BO}} & \frac{\partial \ddot{x}}{\partial F_{CO}} \\ \frac{\partial \ddot{y}}{\partial F_{AO}} & & \vdots \\ \frac{\partial \ddot{z}}{\partial F_{AO}} & \dots & \frac{\partial \ddot{z}}{\partial F_{CO}} \end{bmatrix} = \frac{10^{-8}}{m} \begin{bmatrix} A_x & B_x & C_x \\ A_y & B_y & C_y \\ A_z & B_z & C_z \end{bmatrix}$$

where the unit vectors along the three body fixed axes are defined by

$$\bar{A} = \frac{-\bar{H}(\bar{E} \cdot \bar{H}) + \bar{E}}{|\bar{H}(\bar{E} \cdot \bar{H}) - \bar{E}|} = A_x, A_y, A_z$$

$$\bar{B} = \frac{\bar{E} \times \bar{H}}{|\bar{E} \times \bar{H}|} = B_x, B_y, B_z$$

$$\bar{C} = -\bar{H} = C_x, C_y, C_z$$

and

$\bar{E} = E_x, E_y, E_z$ = Unit vector directed from the probe toward Canopus

$\bar{H} = H_x, H_y, H_z$ = Unit vector directed from the probe toward the Sun

USE

8-4 of 4

CALL BMATRX

The subroutine checks lists in the COMMON area to determine what is required.

CODING INFORMATION

Length of subroutine is 794(10) or 1432(8) words.

REFERENCES

- a. Anderson, John D., RFP 312-37, August 29, 1961.
- b. Null, George W., RFP 312-179, Addendum 3, December 11, 1963.

IDENTIFICATION

9-1 of 2

CALL

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

CALL serves as the logical control routine for the data fitting link, LA6. All steps from reading the ODG data file through computation of partials and residuals to writing the residual file are initiated by this routine.

RESTRICTIONS

a. COMMON break: 46711

b. Subroutines called:

BEER

KINE

COEF

STARP

(DFSB)

REAP

CHADP

DOPLR

COROP

CORR

DICOS

OBTOX

CATS

OFFSYS

WAIT

QUIZ

REJEC

PM360

FORM

GERTA/GREOF

FILL

USE

9-2 of 2

CALL CALL

COMMON input:

IGNAZ data type delete flags (12 × 15)
PTFD-30 data start and stop times
LLIST physical constants solve flags (11)
EPOCH ODP epoch, d. p. sec past 1950.0
THETAI station longitude, deg
SIGX3 rejection sigmas (12 × 15)
DXDR-27 occultation-impact data flag

COMMON output:

HAO observed hour angle from observed optical right ascension
RVECD residual array
TOBL previous observation time

CODING INFORMATION

Length of subroutine is 339 (10) or 528 (8) words.

IDENTIFICATION

10-1 of 5

CATS

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

CATS computes the partials of the ODP data types with respect to the station locations. Such a partial may be non-zero only if the tracking datum is from the same station as the specified station location.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines called:
 - VEC
 - ABDD
 - COS

METHOD

$$\left. \begin{aligned} X_i &= R_i \cos(a_G + \lambda_i) \cos \phi_i \\ Y_i &= R_i \sin(a_G + \lambda_i) \cos \phi_i \\ Z_i &= R_i \sin \phi_i \end{aligned} \right\} \text{(station coordinates)}$$

$$\frac{\partial \rho_i}{\partial R_i} = \frac{1}{R_i} (\underline{R}_i \cdot \underline{L})$$

$$\frac{\partial \rho_i}{\partial \phi_i} = Z_i \left\{ [L_x \cos(a_G + \lambda_i) + L_y \sin(a_G + \lambda_i)] - L_z R_i \cos \phi_i \right\} \left. \vphantom{\frac{\partial \rho_i}{\partial \phi_i}} \right\} \text{(slant range partials)}$$

$$\frac{\partial \rho_i}{\partial \lambda_i} = Y_i L_x - X_i L_y$$

where

10-2 of 5

$$\underline{L} = \frac{\rho + r_s}{\rho}$$

λ_i = station longitude

ϕ_i = station geocentric latitude

R_i = station Earth radius

$$\frac{\partial \dot{\rho}_i}{\partial R_i} = \frac{1}{R_i} (R_i \cdot \underline{L}')$$

$$\frac{\partial \dot{\rho}_i}{\partial \phi_i} = -Z_i \left\{ [L'_x \cos(a_G + \lambda_i) + L'_y \sin(a_G + \lambda_i)] + L'_z R_i \cos \phi_i \right\}$$

$$\frac{\partial \dot{\rho}_i}{\partial \lambda_i} = -Y_i L'_x + X_i L'_y$$

(range
rate
partials)

where

$$L'_x = \frac{1}{\rho_i} (-\dot{x} - 2\omega Y_i + \dot{\rho}_i L_x + \omega y)$$

$$L'_y = \frac{1}{\rho_i} (-\dot{y} + 2\omega X_i + \dot{\rho}_i L_y - \omega x)$$

$$L'_z = \frac{1}{\rho_i} (-\dot{z} + \rho_i L_z)$$

$$\frac{\partial \gamma_i}{\partial R_i} = \frac{-1}{R_i \rho_i} (R_i \cdot \underline{\tilde{D}})$$

$$\frac{\partial \gamma_i}{\partial \phi_i} = \frac{1}{\rho_i} \left\{ Z_i [\tilde{D}_x \cos(a_G + \lambda_i) + \tilde{D}_y \sin(a_G + \lambda_i)] - \tilde{D}_z R_i \cos \phi_i \right\}$$

$$\frac{\partial \gamma_i}{\partial \lambda_i} = \frac{1}{\rho_i} (Y_i \tilde{D}_x - X_i \tilde{D}_y)$$

(elevation
angle
partials)

$$\left. \begin{aligned} \frac{\partial \sigma_i}{\partial R_i} &= \frac{-1}{R_i \rho_i} (R_i \cdot \tilde{A}) \\ \frac{\partial \sigma_i}{\partial \phi_i} &= \frac{1}{\rho_i} \left\{ Z_i [\tilde{A}_x \cos(a_G + \lambda_i) + \tilde{A}_y \sin(a_G + \lambda_i)] - \tilde{A}_z R_i \cos \phi_i \right\} \\ \frac{\partial \sigma_i}{\partial \lambda_i} &= \frac{1}{\rho_i} (Y_i \tilde{A}_x - X_i \tilde{A}_y) \end{aligned} \right\} \begin{array}{l} 10-3 \text{ of } 5 \\ \\ \text{(azimuth} \\ \text{angle} \\ \text{partials)} \end{array}$$

$$\frac{\partial \gamma_i}{\partial S_i} (\tilde{D}) \rightarrow \frac{\partial \delta_i}{\partial S_i} (D) \quad \text{(declination partials)}$$

$$\frac{\partial \sigma_i}{\partial S_i} (\tilde{A}) \rightarrow \frac{\partial a_i}{\partial S_i} (A) \quad \text{(hour angle partials)}$$

where

$$D_x = -\sin \delta_i \cos(a_G + \lambda_i)$$

$$D_y = -\sin \delta_i \sin(a_G + \lambda_i)$$

$$D_z = \cos \delta_i$$

$$A_x = -\sin(a_G + \lambda_i)$$

$$A_y = \cos(a_G + \lambda_i)$$

$$A_z = 0$$

$$\begin{aligned} \tilde{D}_x &= \sin \gamma_i [\sin \sigma_i \sin(a_G + \lambda_i) + \cos \sigma_i \sin \phi_i \cos(a_G + \lambda_i)] \\ &\quad + \cos \phi_i \cos(a_G + \lambda_i) \cos \gamma_i \end{aligned}$$

$$\begin{aligned} \tilde{D}_y &= -\sin \gamma_i [\sin \sigma_i \cos(a_G + \lambda_i) - \cos \sigma_i \sin \phi_i \sin(a_G + \lambda_i)] \\ &\quad + \cos \phi_i \sin(a_G + \lambda_i) \cos \gamma_i \end{aligned}$$

$$\tilde{D}_z = -\cos \sigma_i \cos \phi_i \sin \gamma_i + \sin \phi_i \cos \gamma_i$$

10-4 of 5

$$\tilde{A}_x = -\cos \sigma_i \sin(a_G + \lambda_i) + \sin \sigma_i \sin \phi_i \cos(a_G + \lambda_i)$$

$$\tilde{A}_y = \cos \sigma_i \cos(a_G + \lambda_i) + \sin \sigma_i \sin \phi_i \sin(a_G + \lambda_i)$$

$$\tilde{A}_z = -\sin \sigma_i \cos \phi_i$$

The vectors \underline{A} , \underline{D} , $\tilde{\underline{A}}$, and $\tilde{\underline{D}}$, defined above, are computed by subroutine DICOS.

$$\frac{\partial f_{li}}{\partial S_i} = \frac{\Omega_2}{c} \frac{\partial \dot{\rho}_i}{\partial S_i} \quad (\text{one-way doppler partials})$$

$$\frac{\partial f_{c3i,q}}{\partial S_i} = \frac{\Omega_4}{c} \frac{\partial \dot{\rho}_i}{\partial S_i} \quad (\text{two-way doppler receiver partials})$$

$$\frac{\partial f_{3i,q}}{\partial S_i} = \frac{\Omega_6}{c} \frac{\partial \dot{\rho}_i}{\partial S_i} \quad (\text{three-way doppler receiver partials})$$

$$\frac{\partial f_{dli,j}}{\partial S_i} = \frac{\Omega_2}{c} \frac{\partial \dot{\rho}_i}{\partial S_i} \quad (\text{differenced one-way doppler receiver i partials})$$

$$\frac{\partial f_{c3i,q}}{\partial S_q} = \frac{\Omega_4}{c} \frac{\partial \dot{\rho}_q}{\partial S_q} \quad (\text{two-way doppler transmitter partials})$$

$$\frac{\partial f_{3i,q}}{\partial S_q} = \frac{\Omega_6}{c} \frac{\partial \dot{\rho}_q}{\partial S_q} \quad (\text{three-way doppler transmitter partials})$$

$$\frac{\partial f_{dli,j}}{\partial S_j} = \frac{\Omega_2}{c} \frac{\partial \dot{\rho}_j}{\partial S_j} \quad (\text{differenced one-way doppler receiver j partials})$$

where

Ω_2 = one-way multiplier

Ω_4 = two-way multiplier

Ω_6 = three-way multiplier

USE

10-5 of 5

CALL CATS

COMMON inputs:

RI R_i , km
 PHII ϕ_i , deg
 THETAI λ_i , deg
 GHA α_G , deg
 ELX \underline{L}
 AX \underline{A}
 DX \underline{D}
 TAX $\underline{\tilde{A}}$
 TDX $\underline{\tilde{D}}$
 RVEC ρ_i , km
 OMEGA ω , deg/sec
 X \underline{r} , km
 XDOT $\underline{\dot{r}}$, km/sec
 CZ2 Ω_2
 CZ4 Ω_4
 CZ6 Ω_6

COMMON output:

DIM3 partials $\frac{\partial F}{\partial S_i}$ (12 x 3 x 15)

CODING INFORMATION

Length of subroutine is 421 (10) or 645 (8) words.

REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM33-168, March 18, 1964.
- b. Anderson, John D., RFP 312-37, August 29, 1961.

IDENTIFICATION

11-1 of 5

COEF

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

COEF calculates the coefficients used by the ODP in doppler calculations. The subroutine is cognizant of band differences. All intermediate calculations are in double precision.

RESTRICTIONS

COMMON break: 47675

Subroutines used:

(DFAD)

(DFMP)

(DFSB)

QUIZ

METHOD

$$f_T = f_{T \text{ base}} + \Delta f_T \quad (\text{transponder frequency})$$

$$f_q = f_{q \text{ base}} + \Delta f_q \quad (\text{transmitter frequency})$$

$$\Omega_2 = \omega_2 [f_T + D(t_{ob} - t_0)]$$

$$\Omega_1 = \omega_1 + \Delta\omega_1 - \Omega_2$$

$$\Omega_4 = \omega_4 f_q$$

$$\Omega_3 = \omega_3 + \Delta\omega_3 \text{ for predicts} \quad \Omega_3 = 0 \text{ for data fitting}$$

$$\Omega_6 = \omega_6 f_q$$

$$\Omega_5 = \omega_5 + \Delta\omega_5 - \Omega_6 \text{ for predicts} \quad \Omega_5 = 0 \text{ for data fitting}$$

where for L to S band,

11-2 of 5

$$f_{T \text{ base}} = 2290.0 \times 10^6 \text{ cps}$$

$$f_{q \text{ base}} = 20.00 \times 10^6 \text{ cps}$$

$$k_1 = f_{q \text{ base}} + \Delta k_1$$

$$\omega_2 = 0.3125$$

$$\omega_1 = 9.375 \times 10^6 \text{ cps}$$

$$\omega_4 = 32.579185520$$

$$\omega_3 = 9.375 \times 10^6 \text{ cps}$$

$$\omega_6 = 32.579185520$$

$$\omega_5 = 9.375 \times 10^6 \text{ cps}$$

$$\Delta\omega_1 = 30 k_1$$

$$\Delta\omega_3 = 30 k_1$$

$$\Delta\omega_5 = 30 k_1$$

for L band,

$$f_{T \text{ base}} = 960.0 \times 10^6 \text{ cps}$$

$$f_{q \text{ base}} = 29.66 \times 10^6 \text{ cps}$$

$$\omega_2 = 0.96875$$

$$\omega_1 = 930.15 \times 10^6 \text{ cps}$$

$$\omega_4 = 32.359550561$$

11-3 of 5

$$\omega_3 = 100000. \text{ cps}$$

$$\omega_6 = 31.348314605$$

$$\omega_5 = 930.15 \times 10^6 \text{ cps}$$

$$\Delta\omega_1 = \Delta\omega_3 = \Delta\omega_5 = 0$$

for S band,

$$f_{T \text{ base}} = 2290.0 \times 10^6 \text{ cps}$$

$$f_{q \text{ base}} = 20.00 \times 10^6 \text{ cps}$$

$$k_s = f_{q \text{ base}} + \Delta k_s$$

$$\omega_2 = 1.0$$

$$\omega_1 = 1000000.$$

$$\omega_4 = 104.25339367$$

$$\omega_3 = 1000000.$$

$$\omega_6 = 104.25339367$$

$$\omega_5 = 1000000.$$

$$\Delta\omega_1 = \omega_6 k_s$$

$$\Delta\omega_3 = 0$$

$$\Delta\omega_5 =$$

also

11-4 of 5

D = transponder (beacon) drift

t_{ob} = observation time

t_0 = ODP epoch

USE

CALL COEF

COMMON input:

EP-6 band indication
 EP-7 Δk_1 , cps
 EP-8 Δk_s , cps
 TXCON 6×15 array of $\omega_1, \omega_2, \dots, \omega_6$ coefficients for each station, d. p.
 TFREQ Δf_T , cps
 XFREQ Δf_q , cps
 DXDR-4 D , cps/sec
 TOB observation time, d. p. sec past 1950.0
 EPOCH ODP epoch, d. p. sec past 1950.0
 C10 observed one-way doppler (flag)
 CD10 observed differenced one-way doppler (flag)
 CC30 observed two-way doppler (flag)
 C30 observed three-way doppler (flag)
 CD30 observed DSIF ranging (flag)

COMMON output:

CZ1 Ω_1
 CZ2 Ω_2
 CZ3 Ω_3
 CZ4 Ω_4
 CZ5 Ω_5
 CZ6 Ω_6
 RNJ Ω_5 , d. p.
 PHINJ Ω_6 , d. p.

CODING INFORMATION

11-5 of 5

Length of subroutine is 503 (10) or 767 (8) words.

REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Wollenhaupt, Wilber R., RFP 312-268, April 16, 1964.
- c. Wollenhaupt, Wilber R., RFP 312-319, September 3, 1964.

IDENTIFICATION

12-1 of 3

COL

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

COL calculates the G^2 terms in the weighting function

$$\sigma_j^2 = \sum_{p=1}^6 S_{pjk}^2 G_{pj}^2 \max\left(\frac{t_{pjk}}{\tau}, 1\right)$$

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 - COS
 - SQRT

METHOD

The computation of the g value depends on the value of the weight code group (p) and the data type (j). The following table indicates the g term for each of the p, j combinations:

P Group index		1	2	3	4	5	6
Data type index	j						
ρ_i	1	1	1	$\dot{\rho}_i$	ρ_i	$\Delta_r \rho_i$	1
$\dot{\rho}_i$	2	1	ρ_i	ρ_i	$\Delta_r \dot{\rho}_i$	1	
γ_i	3	1	1	1	$\Delta_r \gamma_i$		
σ_i	4	$\frac{1}{\cos \gamma_i}$	1	1			
δ_i	5	1	1	1	$\Delta_r \delta_i$		
a_i	6	$\frac{1}{\cos \delta_i}$	1	1	$\Delta_r a_i$		
f_{li}	7	Ω_2	Ω_2	$\frac{1}{r}$	$\frac{\Omega_2}{c}$		
f_{c3i}	8	$\frac{\Omega_4}{c}$	$\frac{1}{r}$	$\frac{\Omega_4}{c} \frac{(\rho_i + \rho_q)}{2}$	$\frac{1}{\sqrt{3r}}$	$\frac{\Omega_4}{rc} (\Delta_r \dot{\rho}_i + \Delta_r \dot{\rho}_q)$	$\frac{\Omega_4}{c}$
f_{3i}	9	$\frac{\Omega_6}{c}$	$\frac{1}{r}$	Ω_6	$\frac{\Omega_6}{c} (\Delta_r \dot{\rho}_i + \Delta_r \dot{\rho}_q)$	$\frac{\Omega_6}{c}$	
f_{dli}	10	$\frac{\Omega_2}{c} \rho_i \rho_j$	$\frac{\Omega_2}{c} (\rho_i - \rho_j)$	$\frac{1}{r}$	$\frac{\Omega_2}{c} (\Delta_r \dot{\rho}_i + \Delta_r \dot{\rho}_j)$	$\frac{\Omega_2}{c}$	
ρ_{DSIF}	11	1	1	$\dot{\rho}$	ρ	$\Delta_r \rho$	1

where

12-3 of 3

ρ = slant range
 $\dot{\rho}$ = range rate
 γ = elevation angle
 δ = declination
 $\Delta_r \rho$ = refraction correction to slant range
 $\Delta_r \dot{\rho}$ = refraction correction to range rate
 $\Delta_r \delta$ = refraction correction to declination
 $\Delta_r a$ = refraction correction to hour angle
 Ω_2 = one-way doppler multiplier
 Ω_4 = two-way doppler multiplier
 Ω_6 = three-way doppler multiplier
 τ = doppler averaging time
 c = velocity of light

USE

CALL COL

COMMON input:

RVEC ρ , km
 RDOT $\dot{\rho}$, km/sec
 EL γ , deg
 DEC δ , deg
 DRVEC $\Delta_r \rho$, km
 DRDOT $\Delta_r \dot{\rho}$, km/sec
 DRDEC $\Delta_r \delta$, deg
 DRHA $\Delta_r a$, deg
 CZ2 Ω_2
 CZ4 Ω_4
 CZ6 Ω_6
 TAU τ , sec
 VELC c , km/sec

COMMON output:

GSQ $g^2 (1 \times 6)$

CODING INFORMATION

Length of subroutine is 262 (10) or 406 (8) words

IDENTIFICATION

13

COMAP

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

COMAP checks the mapping requests which are made of the ODP. When step-mapping the product of the U matrices for mapping is computed, thus the U matrix which is used in the calculation of the closest approach parameters is:

$$U = U_{IMP} * U_{T_N} * \dots * U_{T_2} * U_{T_1}$$

where U_{T_N} are the matrices computed for each step-map time and U_{IMP} is evaluated at time of closest approach. Results of the mapping matrix calculations are stored on disk to be printed in a subsequent link.

RESTRICTIONS

- a. ERROR condition: disk error indicated by DC_F
- b. COMMON break: 47055
- c. Subroutines used:

TAMP
PROM
REAM
COMIMP
FLAK
FLOT
DCP
OFFSYS
TYPRYT

USE

This link is called under control of the JPTRAJ Source Deck.

CODING INFORMATION

Length of subroutine is 186(10) or 272(8) words.

IDENTIFICATION

14-1 of 2

COMIMP

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To project the statistical information contained in the covariance matrix $\Gamma(t_0)$ to any later time.

RESTRICTIONS

a. COMMON break: 47055

b. Subroutines used:

MAMUL

PRAMS

(DFAD)

(DFSB)

(DFMP)

METHOD

The procedure employs a mapping matrix

$$U_{t_0, t} = \frac{\partial Q_t}{\partial Q_{t_0}}$$

which is an extended matrix of variational partials representing all estimated parameters Q_{t_0} . If the parameter set Q_{t_0} consists only of the six initial conditions, then this matrix is identical to the familiar U matrix. A similar mapping matrix is employed for including the effect of the considered parameter set \tilde{Q}_{t_0} :

$$V_{t_0, t} = \frac{\partial Q_t}{\partial \tilde{Q}_{t_0}}$$

The mapping operation is then accomplished by

$$\Gamma_t = U \Gamma_{t_0} U^T - V \Gamma_{\tilde{Q}} V^T J^{-1} U^T - U J^{-1} V \Gamma_{\tilde{Q}} V^T + V \Gamma_{\tilde{Q}} V^T$$

If no parameters are being considered, this reduces to

14-2 of 2

$$\Gamma_t = U\Gamma_{t_0}U^T$$

The covariance matrix of estimated parameters is defined as

$$\Gamma = J^{-1} + J^{-1}KJ^{-1}$$

where

$$J^{-1} = (J^* + \tilde{\Gamma}^{-1})^{-1}$$

and

$$K = v\Gamma_{\tilde{Q}}v^T$$

where

$\Gamma_{\tilde{Q}}$ = a priori covariance matrix of the m considered parameters

$$v = \sum_{i=1}^N \phi_i w_i^{-1} \theta_i^T \quad (\theta_i \text{ is the analogous vector of partials of considered parameters})$$

USE

CALL COMIMP

The matrix thus generated is used in the calculation of the target centered covariance matrix (TACCOM) and the closest approach parameters.

CODING INFORMATION

Length of subroutine is 750(10) or 1356(8) words.

REFERENCE

Anderson, John D., RFP 312-37, August 29, 1961.

IDENTIFICATION

15

COMOUT

Michael R. Warner, JPL

IMB 7094 Fap

January 4, 1965

PURPOSE

COMOUT writes the midcourse and trajectory information on the COMENT region of disk.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 COMENT

USE

CALL COMOUT

COMMON input:

- EP-63 midcourse COMENT buffer (29 words forward)
- EP-104 trajectory COMENT buffer (17 words forward)

CODING INFORMATION

Length of subroutine is 14(10) or 16(8) words.

IDENTIFICATION

16-1 of 2

COROP

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

COROP computes the refraction corrections to optically obtained hour angle and declination.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

SQRT

EXP(3)

ARTAN

SIN

METHOD

$$\Delta_r \gamma = \tan^{-1} \left(\frac{A}{\rho - B} \right) \quad (\text{correction to elevation angle})$$

where

$$A = \frac{0.00211}{(\gamma/57.2957795 + 0.0598)^{2.42}}$$

$$B = \sqrt{C^2 - R_e^2 + R_e^2 \sin^2 \gamma} - R_e \sin \gamma$$

$$C = R_e + 51.2064$$

ρ = slant range

R_e = earth equatorial radius

The program then uses $\Delta_r \gamma$ to obtain $\Delta_r \alpha$ and $\Delta_r \delta$ in the same manner as subroutine CORR (q. v.).

USE

16-2 of 2

CALL COROP

COMMON input:

RVEC ρ , km

EL γ , deg

RE R_e , km

COMMON output:

DHA $\Delta_r \alpha$, deg

DDEC $\Delta_r \delta$, deg

CODING INFORMATION

Length of subroutine is 120(10) or 170(8) words.

REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Cain, Dan L., TM 312-275, February 14, 1963.

IDENTIFICATION

17-1 of 3

CORR

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

CORR calculates the refraction corrections and the vertical displacements for angular tracking data.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines called:

SIN

SQRT

METHOD

$$\Delta_r \gamma = 57.2957795 \frac{n}{340.0} b_1 b_2 \quad \gamma < 0.3 \text{ radians}$$

$$\Delta_r \gamma = 57.2957795 n \times 10^{-6} \frac{\cos \gamma}{\sin \gamma} \quad \gamma \geq 0.3 \text{ radians}$$

$$\Delta_r \alpha = \frac{\Delta_r \gamma \cos \phi \sin^2 \alpha}{\cos^2 \gamma \sin \sigma} \quad \delta < 87^\circ$$

$$\Delta_r \delta = \frac{(\sin \phi \cos \gamma - \sin \gamma \cos \phi \cos \sigma) \Delta_r \gamma}{\cos \delta} \quad \delta < 87^\circ$$

$$\Delta_r \alpha = \Delta_r \delta = 0 \quad \delta \geq 87^\circ$$

$$b_1 = 1.0 - (1.216 \times 10^5 b_3 \gamma_{\text{rad}}) - (51.0 - 300.0 \gamma_{\text{rad}}) \sqrt{b_3}$$

$$b_2 = \left[7.0 \times 10^{-4} / (0.0589 + \gamma_{\text{rad}}) \right] - 1.26 \times 10^{-3}$$

$$b_3 = \frac{1}{10^3 (r - R_1)}$$

$$\Delta_v \alpha = \frac{v}{\cos \phi}$$

$$\Delta_v \gamma = u \cos \sigma$$

$$\Delta_v \sigma = u \sin \sigma \frac{\sin \gamma}{\cos \gamma}$$

$$\Delta_v \delta = 0$$

where

- γ = elevation angle ($\gamma_{\text{rad}} = \gamma$ in radians)
- σ = azimuth angle
- α = hour angle
- δ = declination
- Δ_r = refraction correction
- Δ_v = vertical correction
- n = index of refraction (nominally 340.0)
- ϕ = geocentric latitude of station
- r = geocentric probe distance
- R_i = Earth radius at station i
- u = north-south displacement of vertical
- v = east-west displacement of vertical

USE

CALL CORR

COMMON input:

SINEL $\sin \gamma$
 IFA-9 $\neq 0$ to bypass Δ_r and Δ_v calculation
 EL γ
 ELZ $\sin \delta$
 X, Y, Z geocentric probe vector
 RI(i) Earth radius at station i
 FNI(i) index of refraction at station i
 COSEL $\cos \gamma$
 HA α
 COSPHI $\cos \phi$
 SINAZ $\sin \sigma$

COSAZ $\cos \sigma$
 UI(i) north-south displacement, station i
 VI(i) east-west displacement, station i

17-3 of 3

COMMON output:

DEL $\Delta_r \gamma + \Delta_v \gamma$
 DHA $\Delta_r a + \Delta_v a$
 DAZ $\Delta_v \sigma$
 DDEC $\Delta_r \delta$

CODING INFORMATION

Length of subroutine is 188(10) or 274(8) words.

REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Cain, Dan L., IOM to W. Hoover, July 6, 1960.
- c. Cain, Dan L., IOM to W. Hoover, October 4, 1960.

IDENTIFICATION

18-1 of 6

CXPLOT/XXXFSR/CDC

John R. Schoeni, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

To generate on the computer plotting tapes to be displayed on the SC-4020 microfilm recorder. Controlled by a set of pseudo-operations, the routine will scale data, generate and label grid lines, and annotate the plotted information.

RESTRICTIONS

- a. ERROR return when incorrect calling sequence is given.
- b. Subroutine used:
IOCS

USE

CXPLOT uses IOCS to control I/O buffering and the user before calling CXPLOT must open a reserve file for the subroutine's use. In defining the buffer pool to be attached to the plotting file, it is recommended that at least two buffers of up to 200 words in length be assigned. The entry into CXPLOT is from a FAP calling sequence and consists of a call CXPLOT followed by pseudo-operations which specify the operations to be performed. The calling sequence is terminated by an error exit of the form, PZE A, where A is the location of the users error routine. Normal return is to the location following the error return. The following pseudo-operations are recognized.

<p>FVE N, T, D FVE O, O, O</p>	}	Selection of tape unit and indication of film advance
<p>SIX SX, L, SY SIX F, SIZE, D</p>	}	Printing horizontal titles
<p>SVN SX, L, Sy SVN F, SIZE, D</p>	}	Printing vertical titles
<p>PON Rx₀, Rx_n PON Ry₀, Ry_n PON n, e, p PON m, e, q PON Fx, Fy PON X₀, X_n PON Y₀, Y_n</p>	}	Generation and labelling of grid

PTW Lx ₀ , EX, Ly ₀ } PTW N, A, CHAR }	Generation of graph	18-2 of 6
PTH X, t, Y	Generation of special grid line or form flash	
MZE O, t, N	Termination of plot	
PZE A,	Error return	

CONTROL PSEUDO-OPERATIONS

FVE N, T, D

An FVE pseudo-operation must be the first pseudo-operation encountered in the first calling sequence to CXPLOT in a program because it causes various addresses throughout the subroutine to be initialized.

N In the FVE pseudo-operation, N indicates that every Nth point will be plotted.

N = 0 = 1

T T = 1 will cause a film advance command to be generated.

T = 0 indicates no frame advance.

D is the symbolic location of the File Control Block that has been opened for this file.

FVE O, O, O

A second FVE pseudo-operation is used when the output is to be written on the IBM 7094 using a packed format and high density tape mode. The plot tape generated by the use of this option requires processing on the 1401 before being plotted on the SC-4020. The Standard SFOF mode of operation is high density and packed.

MZE O, T, N

The MZE pseudo-operation gives information required to terminate a particular plot or set of plots.

If N is zero, the instruction is treated as a skip, and control will pass to the next pseudo-operation.

If N is non-zero, STOP PRINT and ADVANCE FRAME commands will be generated and written on the tape previously specified.

T not used.

PZE A

The PZE pseudo-operation indicates the end of the calling sequence. Normal exit is to the instruction following this pseudo-operation. "A" is the address of a user supplied error routine.

HORIZONTAL TITLES

18-3 of 6

SIX SX, L, SYSIX F, SIZE, D

The SIX pseudo-operations give information required for printing horizontal titles. The first operation in the pair describes where the title is to be located. The second operation tells the location of the format statement and the letter size. Six pseudo-operations must always occur in pairs.

L = 1 SX and SY designate the X and Y coordinates locating the center of the title.

L = 0 SX and SY designate the X and Y coordinates locating the position at which the first character of the title is to be centered.

F designates the location of the format statement to be used for the title. The format statement starting at location F must be of the form

(xHyz -----).

x an integer indicating the number of hollerith characters which follow the H.

y if blank the current frame will be used. This is the normal case.

y if equal to 1, a film advance command is generated before printing this title.

z is always a blank character.

SIZE designates the letter size and may have the values 0, 1, 2, 3 specifying that letters of 1, 2, 4, or 8 times normal size are to be used. (Caution: a character 8 times normal size is 64 by 128 units in size.)

D When printing normal size letters, SIZE is zero, D indicates whether the letters will be generated by the vector generator or by the plotting matrix.

D = 0: plotting matrix.

D = 1: vector generator.

VERTICAL TITLES

SVN SX, L, SYSVN F, SIZE, D

The SVN pseudo-operations give information required for printing titles vertically. They are analogous to the SIX pseudo-operations in format.

GRIDS & LABELS

PON

The PON pseudo-operations specify the grid lines to be drawn, the exposure to be used in drawing them, the formats to control their labelling and the scaling to be used in the plot.

The first five PON pseudo-operations must be used as a group.

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PON Rx_0, Rx_n
 PON Ry_0, Ry_n
 PON n, e, p
 PON m, e, q
 PON Fx, Fy

PON Rx_0, Rx_n

Rx_0 is the location of the lower end of the range of x values.

Rx_n is the location of the upper end of the range of x values.

The limits for the range will be in floating point form.

PON Ry_0, Ry_n

is analogous to the PON Rx_0, Rx_n for the range of y.

PON n, e, p

contains an integer n specifying the number of uniform intervals in the x-direction (vertical grids) to be bounded by grid lines, and an integer p indicating that the first grid line and every pth grid line following will be labelled. If e = 1, the labelled grid lines will be drawn heavy; otherwise they will be light (vertical grids).

PON m, e, q

This is analogous to the third PON but refers to the y-direction (horizontal grids).

PON Fx, Fy

contains the locations of format statements to be used in labelling the horizontal and vertical grid lines, respectively. These should be of the form:

FX BCI , (F c. d),

where

F indicates that a conversion from a floating point number to fixed point number should be performed, and that the fixed point number should be printed.

c designates the "column width." ($c \leq 9$)

d designates the number of digits to be retained to the right of the decimal point.
 $d < (c-1)$

PON X_0, X_n

PON Y_0, Y_n

The sixth and seventh PON pseudo-operations specify the area of the character on surface (1023 x 1023) to be used for plotting. These two PON pseudo-operations may be omitted, and if so the standard case of

PON 96,, 992
 PON 0,, 896

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will be assumed by the subroutine.

PON X₀, X_n

is the range of the horizontal positions to be used

$$(0 \leq X_0 < X_n \leq 1023).$$

PON Y₀, Y_n

contains the range of vertical positions to be used for the graph

$$(0 \leq Y_0 < Y_n \leq 1023).$$

GENERATION OF GRAPH

PTW Lx₀, EX, Ly₀

PTW N, A, CHAR

The PTW pseudo-operations specify the character and the exposure to be used in plotting, the memory area containing the sets of (x, y) values to be plotted, and the scaling factors to be used.

Lx₀ designates the starting memory location for the sequence of values of x to be plotted.

Ly₀ designates the starting location for the corresponding values of y.

N designates the number of pairs of values of x and y to be plotted.

Thus, locations Lx₀ to Lx₀+N - 1 contain the values of x corresponding to the values of y found in locations Ly₀ to Ly₀+N - 1.

Note the special case where the data are in x, y pairs. N is the number of x and y values, and the address of the FVE pseudo-operation is set to 2. The same type of logic applies to triplets, etc.

EX designates the exposure to be used in plotting. If EX = 0, the exposure will be heavy; otherwise, the light mode will be used.

A=0 The scaling factors computed from the previous set of PON pseudo-operations will be used.

A≠0 The four locations following the pair of PTW instructions should contain the location of floating point values of a, b, c, and d in that order.

For each pair of values, x, y, a corresponding pair of coordinates (X, Y) is computed by the subroutine as follows:

$$X = ax + b$$

$$Y = cy + d$$

where

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a, b, c and d are the scaling parameters. They depend upon the range of coordinates to be used when plotting the range of values specified.

If, in scaling, a point is either $0 > X > 1024$ or $0 > Y > 1024$, the point is discarded.

CHAR The Hollerith character to be used for the plot is given in the six bit positions 30 to 35 of location CHAR.

JOINING POINTS BY VECTORS

If the character specified is the space the vector generator will be used for the plot, and successive points will be connected using straight line segments.

PTH X, t, Y

The PTH pseudo-operation allows the programmer two options: one to draw grid lines directly in either the x or y direction, and one to control usage of the form flash.

t = 0 A horizontal grid line will be drawn beginning at X, Y and extending to the right edge of the frame.

t = 1 A vertical grid line will be drawn from X, Y to the upper edge of the frame.

t = 4 A form flash command will be generated. Care must be taken not to generate more than one form flash on the same plot.

CODING INFORMATION

Length of subroutine is 1744(10) or 3320(8) words.

IDENTIFICATION

19-1 of 2

DATAPE

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To generate a simulated data file for checking and for study programs.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

IXTAB
 OFFSYS
 (DFAD)
 TIMER
 LOOKUP
 INTR1
 GHADP
 COEF
 DOPLR
 CORR
 ND2F
 FILT

Types of doppler are limited as follows:

If IX (transmitter) = 0, C3 and CC3 are zero.

If IX ≠ 0, IR (receiver) ≠ IX, C1 and CC3 are zero.

If IX ≠ 0, IR = IX, C1 and C3 are zero.

METHOD

Data identified under the control card DATA TAPE SIGMA will be included on the file.

Weights for each data type must be included under WEIGHTS BY DATA TYPE AND STATION. The equation for the computation of the data value in the file is:

$$R(I) = R_c(I) + \Delta R(I) + F(I, IR) * A + B(I, IR)$$

where

R(I) = the array of data where I references the data type

R_c(I) = calculated value of the data point

$\Delta R(I)$ = correction on calculated value

19-2 of 2

$F(I, IR)$ = noise factor on data by type and station. This provided by setting the data type under DATA TAPE SIGMA to a floating point number.

A = random number

$B(I, IR)$ = bias. Obtained by value under DATA TAPE BIAS according type.

Data is calculated by the same routines employed in the fitting of data.

USE

CALL DATAPE

CODING INFORMATION

Length of subroutine is 301(10) or 455(8) words.

IDENTIFICATION

20

DECOD

Michael R. Warner, JPL
 IBM 7094 Fap
 January 4, 1965

PURPOSE

DECOD determines the group value for the given group number of the weight code word. This word consists of six 3-bit groups from bit 5 through bit 17. Each group may have a value from 0 through 7. DECOD also floats the data sample rate, which occupies the address portion of the weight code word.

RESTRICTION

COMMON break: 47675

USE

CALL DECOD

PZE J group number

PZE K group value

COMMON input:

RVECW weight code array (1 × 11)

ID data type identification

COMMON output:

TS sample rate, sec.

CODING INFORMATION

Length of subroutine is 32 (10) or 40 (8) words.

IDENTIFICATION

21

DIAG/DIAGO

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To compute standard deviation of the mapped forward covariance matrix or standard deviation of input covariance matrix.

RESTRICTIONS

- a. COMMON break: 47675
- b. Each subroutine computes the square root of the diagonal of a specific matrix.
- c. Subroutines used:
 SQRT

USE

CALL DIAG Standard deviation of input covariance matrix

CALL DIAGO Standard deviation of mapped forward covariance matrix

CODING INFORMATION

Length of each subroutine is 105 (10) or 151 (8) words.

IDENTIFICATION

22

DICOS

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

DICOS computes the direction cosine vectors \underline{A} , \underline{D} , $\tilde{\underline{A}}$, and $\tilde{\underline{D}}$, which are defined in the documentation of subroutine CATS.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
SIN/COS

USE

CALL DICOS

COMMON input:

DEC	δ , deg
GHA	α_G , deg
THETA1	λ_1 , deg
EL	γ , deg
AZI	σ , deg
PHI1	ϕ_1 , deg

COMMON output:

AX	\underline{A}
DX	\underline{D}
TAX	$\tilde{\underline{A}}$
TDX	$\tilde{\underline{D}}$

CODING INFORMATION

Length of subroutine is 123 (10) or 173 (8) words.

IDENTIFICATION

23-1 of 4

(DISCBU) REDE/RITE/PISA
Charles Coltharp, JPL
IBM 7094 Fap
January 4, 1965

PURPOSE

DISCBU provides elementary buffering for reading and writing scratch information on disk.

RESTRICTIONS

Subroutines used:
DCP

METHOD

DISCBU uses DCP for reading and writing the disk. All physical records are 200 words in length. Each logical record is preceded on the disk by a control word,

PZE O, ,L

where L is the number of words in the logical record. L may be zero, but L will never be greater than 200.

A control word of

MZE O, ,O

denotes a logical end of file.

USE

To write	CALL	RITE	
	PZE	FILE, ,	ERREOF
	IOXY	A, ,	N

To read CALL REDE
PZE FILE,, ERREOF
IOXY A,, N
.
.
.

23-2 of 4

A is the location into which information is read, or from which it is written.

TCH, transfer in channel, will be recognized by both read and write operations.

The tag portion of all IO commands is ignored.

The address and decrement portions of IOCD commands are ignored.

Closed or unopened files are opened automatically by the first read or write operation requested.

Read Operation

- IOCD Rewind file; close file. When IOCD is used it must be the only command in the command list.
- IOCP Read N words, ignoring logical record marks, and proceed to next command. If this command precisely finishes a record, the logical end of record may be recognized by the next command.
- IOCT Read N words, ignoring logical record marks, and terminate IO. Unread words in a partially read record will be lost.
- IORP Read to logical record mark (read one whole record, or finish a record already started), and proceed to next command. Put number of words read into the decrement portion of the IORP command. N is not interpreted.
- IORT Same as IORP, but terminate IO.
- IOSP Same as IORP if the number of words in the record is less than or equal to N. Otherwise, transmission is stopped after N words have been read. Unread words in this record will be lost. Control proceeds to the next command.
- IOST Same as IOSP, but terminate IO.

Write Operation

- IOCD Write logical EOF, rewind file; close file. When IOCD is used it must be the only command in the command list.
- IOCP Write N words, and proceed to next command.
- IOCT Write N words followed by a logical record mark, and terminate IO
- IORP Write N words followed by a logical record mark, and proceed to next command.

IORT Same as IOCT.

23-3 of 4

IOSP Same as IORP.

IOST Same as IORT.

If at any time in writing, a record reaches a length of 200 words before a logical record mark is indicated by the user, DISCBU will insert a logical record mark before continuing the transmission.

File Block

```
FILE  BCI  1,  XXXXXX
      +1  PZE  J  (decrement = P)
      +2  PZE  K  (decrement = Q)
      +3  PZE  BUFFER (decrement = R)
      +4  PZE  200
```

XXXXXX an entry in the disk directory, defined as scratch

J that record in XXXXXX with which this file begins

K that record in XXXXXX with which this file ends

BUFFER the location of a 400 word block of core storage

The decrements of FILE +1 through FILE +3 (the items in parentheses, above) will be used by DISCBU for bookkeeping. These items will be zero when the file is not open, and they should be set zero initially by the user.

P last disk record transmitted

Q buffer designation

0 = next disk transmission will be from (or into) BUFFER + 0

1 = next disk transmission will be from (or into) BUFFER + 200

R location of next word in BUFFER to be read or written.

FILE +4 is the location given to DCP for the flag word. Initially this must be a positive, non-zero number.

Error - End of File Return

Control is returned to ERREOF under the conditions described below.

If an error is indicated by DCP, that indication will be in the AC on return. The error codes are described in the design specifications for DCP.

If the user tries to exceed the self-imposed file bounds (K is the largest usable disk record in XXXXXX), the AC on return will be MZE 0,0,-1.

No provision is made for continuing if the condition occurs on a read operation.

Before the file may be used again, the user must issue the sequence

```
CALL  REDE
PZE   FILE,,  ERREOF
IOCD
```

to close and rewind the file.

23-4 of 4

If the condition occurs on a write operation, the user may continue, provided all seven index registers are preserved, by executing

TRA $\$PISA$.

However, K should be increased if the user wishes to read the material back at a later time.

If a logical EOF is encountered by a read operation, the AC will be plus zero.

CODING INFORMATION

Length of subroutine is 273 (10) or 421 (8) words.

IDENTIFICATION

24

DNAME

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

DNAME contains a list of the BCD names of the eleven ODP data types.

USE

The BCD names, one per word, are stored backward. The entry point, DNAME, is assigned to the first logical location (last physical location).

CODING INFORMATION

Length of subroutine is 12 (10) or 14 (8) words.

IDENTIFICATION

25-1 of 5

DOPLR
 Michael R. Warner, JPL
 Fortran II, Version 3
 January 4, 1965

PURPOSE

DOPLR obtains the calculated values of the ODP data types.

RESTRICTIONS

COMMON break: 47675

Subroutines called:

ABDD
 ARSIN
 ARTAN
 DMOD
 RATES
 SQRT
 VEC
 (DFAD)
 (DFMP)

METHOD

DOPLR first obtains the fine light-time correction in a two-iteration procedure:

$$\epsilon = 0$$

$$\underline{r}_t = \underline{r}_t' + \epsilon \underline{\dot{r}}_t'$$

$$\underline{\dot{r}}_t = \underline{\dot{r}}_t' + \epsilon \underline{\ddot{r}}_t'$$

$$\underline{\ddot{r}}_t = \underline{\ddot{r}}_t' + \epsilon \underline{\overset{\cdot\cdot}{r}}_t'$$

$$\underline{\overset{\cdot\cdot}{r}}_t = \underline{\overset{\cdot\cdot}{r}}_t'$$

$$\Delta t = \frac{\rho + \Delta t' \dot{\rho}}{C + \dot{\rho}}$$

$\epsilon = \Delta t' - \Delta t$; the procedure is then repeated.

where

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$\Delta t'$ = coarse correction from subroutine TIMER

t' = $t_{ob} - \Delta t'$

ρ = slant range at t'

$\dot{\rho}$ = range rate at t'

c = speed of light

The eleven ODP data types are then obtained. DOPLR calls subroutine RATES to obtain slant range and its derivatives and all other leg-dependent quantities:

$$\rho = \left| \underline{r}_t - \underline{R}_{it_{ob}} \right| \quad (\text{slant range})$$

where

\underline{r} = geocentric position vector of probe

\underline{R}_i = geocentric position vector of station i

$$\dot{\rho} = \frac{\underline{\rho} \cdot \dot{\underline{\rho}}}{\rho} \quad (\text{range rate})$$

where

$\underline{\rho}$ = topocentric position vector of probe

$\dot{\underline{\rho}}$ = topocentric velocity vector of probe

$$\gamma = \sin^{-1} \frac{\underline{R}_i \cdot \underline{L}}{R_i} \quad (\text{elevation angle})$$

where

$$\underline{L} = \frac{\underline{\rho} + \underline{r}_s}{\rho}$$

R_i = Earth radius at station i

\underline{r}_s = geocentric position vector of Sun

$$\sigma = \tan^{-1} \frac{\sin \gamma}{\cos \gamma} \quad (\text{azimuth angle}).$$

where

25-3 of 5

$$\sin \sigma = \frac{L_y \cos(a_G + \lambda_i) - L_x \sin(a_G + \lambda_i)}{\cos \lambda_i}$$

$$\cos \sigma = \frac{-L_x \sin \phi_i \cos(a_G + \lambda_i) - L_y \sin(a_G + \lambda_i) \sin \phi_i + L_z \cos \phi_i}{\cos \gamma_i}$$

a_G = Greenwich hour angle

λ_i = longitude of station i

ϕ_i = geocentric latitude of station i

$$a = a_G + \lambda_i - \tan^{-1} \left(\frac{\rho_y}{\rho_x} \right) \quad \text{(hour angle)}$$

$$\delta = \sin^{-1} L_z \quad \text{(declination)}$$

$$f_1 = \Omega_1 + \Omega_2 \phi_1 \quad \text{(one-way integrated doppler frequency)}$$

$$f_{c3} = \Omega_3 + \Omega_4 \phi_3 \quad \text{(coherent three-way integrated doppler frequency)}$$

$$f_3 = \Omega_5 + \Omega_6 \phi_3 \quad \text{(three-way integrated doppler frequency)}$$

where

$\Omega_1, \Omega_2, \dots, \Omega_6$ are defined in COEF writeup.

$$\phi_1 = \frac{\dot{\rho}_i}{c} - \frac{1}{c^2} h_1 - \frac{\ddot{\rho}_i}{c} \frac{\tau^2}{24}$$

$$\phi_3 = \frac{1}{c} (\dot{\rho}_i + \dot{\rho}_q) - \frac{1}{c^2} h_3 - \frac{1}{c} (\ddot{\rho}_i + \ddot{\rho}_q) \frac{\tau^2}{24}$$

τ = doppler counting interval

q = transmitter index

i = receiver index

$$h_1 = \frac{\dot{\rho}}{\rho} \underline{\rho} \cdot (\underline{r} - \underline{r}_s) + \frac{1}{2} \left(\dot{R}_{it_{ob}} - \dot{r}_t^2 \right)$$

$$h_3 = \dot{\rho}_i^2 + \dot{\rho}_q \dot{\rho}_i + \frac{\dot{\rho}_i}{\rho_i} \underline{\rho}_i \cdot \left(\dot{R}_{it_{ob}} - \dot{r}_{st} \right)$$

$$- \frac{\dot{\rho}_q}{\rho_q} \underline{\rho}_q \cdot \left(\dot{R}_{qt_{tr}} - \dot{r}_{st} \right) + \frac{1}{2} \left(\dot{R}_{it_{ob}}^2 - \dot{R}_{qt_{tr}}^2 \right)$$

$$f_{dl} = f_{li} - f_{lj} \quad \text{(differenced one-way doppler from stations i and j)}$$

where f_{li} and f_{lj} are one-way doppler values which must be taken simultaneously at stations i and j

$$\rho_{DSIF} = \frac{\Omega_6(\rho_q + \rho_i)}{16c} + \epsilon \quad \text{(DSIF ranging)}$$

where

ϵ = ranging system bias

USE

CALL DOPLR

COMMON input:

DELTA	Δt^1 , d. p. sec past 1950.0
X	$\underline{r}(1 \times 3)$, km
XDOT	$\underline{\dot{r}}(1 \times 3)$, km/sec
XAC	$\underline{\ddot{r}}(1 \times 3)$, km/sec ²
XJERK	$\underline{\dddot{r}}(1 \times 3)$, km/sec ³
VELC	c, km/sec
CZ1, . . . CZ6	$\Omega_1, \dots, \Omega_6$
RVECO	observed values of ODP data types (1 x 11)
OMEGA	Earth rotation rate, deg/sec

COMMON output:

RVEC	calculated values of ODP data types (1 x 11)
DELTA2	Δt , d. p. sec past 1950.0

CODING INFORMATION

25-5 of 5

Length of subroutine is 715 (10) or 1313 (8) words.

REFERENCE

Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.

IDENTIFICATION

26-1 of 2

ENDIT

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

ENDIT sets up and solves the normal equations. The covariance matrix

$$\Gamma = J^{-1} + J^{-1} KJ^{-1}$$

is obtained from the matrices accumulated by subroutine FORM. The ΔQ vector is retained for subsequent output.

RESTRICTIONS

COMMON break: 47675

Subroutines used:

(DFSB)/(DFMP)/(DFAD)

PRIM

SQRT

STPREG

REVRT

NOUT

METHOD

The matrix inversion is accomplished by subroutine STPREG (q. v.).

USE

CALL ENDIT

COMMON input:

KLST attitude control estimate flags

FABC attitude control coefficients

ILIST	initial conditions estimate flags	26-2 of 2
XNJ	initial conditions	
LLIST	physical constants estimate flags	
GRAVE	physical constants	
MLIST	velocity of light estimate flag	
VELC	velocity of light	
NLIST	station locations estimate flags	
RI	station locations	
GFLIP	a priori Γ^{-1}	
QTILD	nominal values corresponding to QTILD	
NEST	number of estimated parameters	
NCON	number of considered parameters	
RIGHT	accumulated right-hand side	
XJ	accumulated normal matrix	
SOUR	sum of squares	
NTOT	number of data points	
XX	accumulated consider matrix	
XMU	accumulated consider vector	

COMMON output: .

QNU	updated parameter list
RIGHT	ΔQ vector
XJ	J inverse
CPGAM	covariance matrix
XNJ	} updated parameters
GRAVE	
RI	
FABC	
CAPQ	Q sum of squares

CODING INFORMATION

Length of subroutine is 828 (10) or 1474 (8) words.

IDENTIFICATION

27

EPPHEM

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To prepare an ephemeris tape with complete information required from the planetary ephemeris and the probe ephemeris for each time point of the ODP. This program checks the requirements of each iteration and interpolates and if necessary integrates the partials of the physical constants. The information is written on tape A-4.

RESTRICTIONS

a. COMMON break: 47055

b. Subroutines used:

GAMAT

LOOKUP

RITEM

TOCIM

INTR1

PONT

STARP

WREOF

KINE

QUIZ

TIMER

(DFSB)

METHOD

EPPHEM is primarily a logical decision box, which accomplishes its task through the use of subroutines. A flow chart is given to supply further clarification.

USE

CALL EPPHEM

CODING INFORMATION

Length of subroutine is 361 (10) or 551 (8) words.

IDENTIFICATION

28-1 of 2

ERROR

Michael R. Warner, JPL
 IBM 7094 Fap
 January 4, 1965

PURPOSE

ERROR prints ODP error condition messages. There are three versions of this subroutine in the ODP; the version in link LA9 prints recoverable error condition messages, the version in link LA12 prints unrecoverable error messages, and the version in the other links stores the calling sequence for subsequent action by LA12.

RESTRICTIONS

COMMON break: 47675
 Subroutines used: (LA9)
 TYPRYT
 ONLIN
 Subroutines used: (LA12)
 TYPRYT
 PROUT
 ENDSYS
 RECOV

USE

CALL ERROR
 PZE I }
 PZE J } Decrement integers
 PZE K }

<u>I</u>	<u>Error message</u>	<u>Version</u>
1	INCORRECT RECORD NUMBER IN DISK CALLING SEQUENCE	LA12
2	WEIGHT (J = ID) FOR STATION (K = IR) = 0	LA12
3	DISK STORAGE NOT ALLOCATED FOR THIS JOB	LA9

4	ESTIMATE OR CONSIDER LIST EXCEEDS 20 PARAMETERS	LA9
5	NO INPUT RECEIVED WITHIN TIME LIMIT	LA9
6	PROGRAM UNABLE TO READ/WRITE PROBE EPHEMERIS	LA12
7	FIRST PROBE EPHEMERIS TIME GREATER THAN LOOKUP ARGUMENT	LA12
8	LAST PROBE EPHEMERIS TIME LESS THAN LOOKUP ARGUMENT	LA12
9	INJECTION CONDITION = 0	LA9
10	INPUT J MATRIX NOT N*N	LA9
11	INADMISSIBLE SYMBOL OR DATA FIELD	LA9
12	INPUT CONSIDER MATRIX NOT M*M	LA9
13	POINTING TIME LESS THAN EPOCH + R/C	LA9
14	MAPPING TIME LESS THAN EPOCH	LA9
15	DISK TRANSMISSION ERROR	LA12, LA9
16	DISK BUFFER ALLOCATION INSUFFICIENT	LA12
17	PLANETARY EPHEMERIS ERROR	LA12
18	DATA POINT OUT OF TIME SORT	LA12
19	ERROR IN CXPLOT CALLING SEQUENCE	LA12
20	ERROR IN DP SQRT SUBROUTINE	LA12, LA9
21	LOOKUP TIME FAILS TO MATCH TIME ON EPHEMERIS TAPE	LA12

The second line of messages 1, 2, 3, 6, 7, 8, 15, 16, 17, 18, 19, 20, 21, is
RECOVERY IMPOSSIBLE. JOB ABORTED.

The second line of messages 4, 5, 9, 11, 13, 14, is
USER MUST CORRECT AND RELOAD INPUT DECK WITHIN 3 MINUTES.

The second line of messages 10, 12, is
CONDITION IGNORED. JOB WILL CONTINUE.

CODING INFORMATION

Length of the LA9 version is 286 (10) or 436 (8) words.

Length of the LA12 version is 326 (10) or 506 (8) words.

Length of the dummy version is 9 (10) or 11 (8) words.

IDENTIFICATION

29

FDATA

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

FDATA stores the nominal values of the S and T weighting tables in COMMON arrays SSQ and TL, respectively. The S values are squared prior to their storage in SSQ.

RESTRICTIONS

COMMON break: 47675

USE

CALL FDATA

CODING INFORMATION

Length of subroutine is 591 (10) or 1117 (8) words.

REFERENCE

Hamilton, Thomas W., Inter-Office Memorandum to M. Warner, June 1962.

IDENTIFICATION

30-1 of 2

FILL/FILT

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

FILL writes the ODP residual file on disk in the same physical format as the ODG-generated data file. It is also used to write the tracking predictions on disk for subsequent output. FILT writes a simulated ODG data file on disk. Both entries use the RITE entry of the buffered disk routine DISCBU.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 - RITE
 - PISA

USE

CALL FILL

PZE

CALL FILT

PZE

COMMON input:

TOB	time of observation, d. p. sec past 1950.0
RVEC	array of computed data types (1 × 11)
RVECD	array of residuals (1 × 11)
RVECW	array of weights (1 × 11)
IF5	≠ 0 normal entry, = 0 end-of-file entry
EP-6	band designation
XFREQ	transmitter frequency
TFREQ	transponder frequency
IR	receiver ID, B17
IX	transmitter ID, B17
TAU	doppler averaging time, sec

COMMON output:

30-2 of 2

TAB-799 400 word output buffer for DISCBU

The disk logical record format is as follows:

Word	0	word count (2N + 5, B17) or - 0 for EOF
	1	TOB
	2	TOB-1
	3	XFREQ if IX ≠ 0 TFREQ if IX = 0
	4	0
	5	<u>bit</u> S 1 if L - S band 0 if L or S band
		1 1 if RVEC ≠ 0
		2 1 if RVEC - 1 ≠ 0
		3 1 if RVEC - 2 ≠ 0
		· ·
		· ·
		· ·
	11	1 if RVEC - 10 ≠ 0
	12-15	not used
	16	1 if L - S or S band 0 if L band
	17	not used
	18-21	IX
	22-25	IR
	26-35	TAU
	6	first member of RVEC which is non-zero
	7	<u>bits</u> S-17 RVECD corresponding to word 6 18-35 RVECW corresponding to word 6
	8	second member of RVEC which is non-zero
	9	second RVECD - RVECW word
		·
		·
		·
	2N + 4	Nth member of RVEC which is non-zero
	2N + 5	Nth RVECD - RVECW word

CODING INFORMATION

Length of subroutine is 191 (10) or 277 (8) words.

IDENTIFICATION

31

FIRST

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

Main program for link 1 of SFOF-JPTRAJ version of the JPL Orbit Determination Program. FIRST clears COMMON and formats the rejection table on the disk.

RESTRICTIONS

- a. ERROR condition: disk error indicated by DCP
- b. COMMON break: 46711
- c. Subroutines used:
 - FDATA
 - NOMNL
 - FLAK
 - DISCBU

USE

Initiates ODP by clearing COMMON and calling subroutines to store the permanent or semi-permanent information. This link is called only once per "run" of the ODP.

CODING INFORMATION

Length of subroutine is 427 (10) or 653 (8) words.

IDENTIFICATION

32

FIT

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

Main program for link 6. The purpose of the routine is to control calculation of the observables and the linkage.

RESTRICTIONS

- a. ERROR condition: disk error indicated by DCP
- b. COMMON break: 46711
- c. Subroutines used:
 - CALL
 - OCIM
 - FLAK
 - DISCBU
 - TAPIO

USE

When fitting data, this link is called by the JPTRAJ monitor.

CODING INFORMATION

Length of subroutine is 476 (10) or 734 (8) words.

IDENTIFICATION

33

FLAT/FLAPR/FLAK

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

Subroutine to check the I/O activity flags.

RESTRICTIONS

a. ERROR exit to indicate disk I/O error

b. Subroutines used:

ERROR

TYPRYT

USE

CALL FLAT	To check TAPIO flag
CALL FLAPR	To check PROUT flag
CALL FLAK	To identify disk error return

CODING INFORMATION

Length of subroutine is 64 (10) or 100 (8) words.

REFERENCE

EPD - 125, Rev. 1, April 1, 1964

IDENTIFICATION

34-1 of 2

FORM

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

FORM sets up the column matrix of partials $\partial F/\partial Q$ and accumulates the normal matrix J.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 - PARLEY
 - (DFMP)/(DFAD)/(DFSB)
 - SQRT

METHOD

The partials $\partial F/\partial Q$ are obtained by the chain rule when Q is an initial condition or physical constant:

$$\frac{\partial F}{\partial Q} = \frac{\partial F}{\partial \underline{r}} \frac{\partial \underline{r}}{\partial Q}$$

If Q is the velocity of light or a station parameter, $\partial F/\partial Q$ are computed analytically and input to FORM.

USE

- A = 1.0 to execute entire routine
- CALL FORM
- A = -1.0 to bypass chain rule section (occultation or impact time partials)
- CALL FORM

COMMON input:

- RVECW weight for data type ID
- ID data type identification
- POBX partials $\partial F/\partial \underline{r}$ (6 × 12)
- GMAT partials $\partial \underline{r}/\partial q$ (6 × 11)
- U partials $\partial \underline{r}/\partial r_0$ (6 × 6)
- LLIST initial conditions flags (6)
- LLIST physical constants flags (11)

MLIST	velocity of light flag (1)	34-2 of 2
NLIST	station locations flags (3 x 15)	
QA	partials $\partial F/\partial q$ from occultation-impact	
QB	partials $\partial F/\partial r_0$ from occultation-impact	
DRDC	partials $\partial F/\partial c$	
DIM3	partials $\partial F/\partial S_i$	
NEST	number of estimated parameters	
NCON	number of considered parameters	
RVECD	residual, data type ID	
IF10	phi vector output flag	
DXDR-13	MMP output flag	

COMMON output:

CPPHI	ordered partials $\partial F/\partial Q$
CPTHT	ordered partials $\partial F/\partial \tilde{Q}$ (considered)
XJ	J matrix
RIGHT	right-hand side of normal equations
XX	J matrix for considered parameters
NTOT	number of data points
SOUR	sum of squares

CODING INFORMATION

Length of subroutine is 566 (10) or 1066 (8) words.

IDENTIFICATION

35

G1G2

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

G1G2 computes the gravity equations g_1 and g_2 .

RESTRICTION

COMMON break: 47675

METHOD

$$\begin{aligned}
 g_1 = 1 + & \left[3 - 42 \left(\frac{z}{r} \right)^2 + 63 \left(\frac{z}{r} \right)^4 \right] \left[\frac{D}{7} \left(\frac{R_e}{r} \right)^4 \right] + \left[H \left(\frac{R_e}{r} \right)^3 \right] \left[3 - 7 \left(\frac{z}{r} \right)^2 \right] \frac{z}{r} \\
 & + \left[J \left(\frac{R_e}{r} \right)^2 \right] \left[1 - 5 \left(\frac{z}{r} \right)^2 \right] \\
 g_2 = 1 + & \left[15 - 70 \left(\frac{z}{r} \right)^2 + 63 \left(\frac{z}{r} \right)^4 \right] \left[\frac{D}{7} \left(\frac{R_e}{r} \right)^4 \right] - \left[H \left(\frac{R_e}{r} \right)^3 \right] \left[1 - 10 \left(\frac{z}{r} \right)^2 \right] \\
 & + \frac{35}{3} \left(\frac{z}{r} \right)^4 \frac{3}{5} \frac{r}{z} + \left[J \left(\frac{R_e}{r} \right)^2 \right] \left[3 - 5 \left(\frac{z}{r} \right)^2 \right]
 \end{aligned}$$

USE

CALL G1G2 (G1, G2) Results are stored in G1 and G2.

CODING INFORMATION

Length of subroutine is 182 (10) or 266 (8) words.

REFERENCE

Anderson, John D., TM 312-131, August 23, 1961.

IDENTIFICATION

36-1 of 4

GAMAT

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To obtain the partials $\partial r/\partial q$ from the following integral using Simpson's rule:

$$\gamma(t) = U(t) \int_{t_0}^{t_U^{-1}} (t^*)B(t^*) dt^*$$

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

BMATRIX

INSPC

INTR1

LOOKUP

(DFAD)

(DFDP)

(DFMP)

(DFSB)

METHOD

The partials of acceleration with respect to physical constants $\partial \ddot{r}/\partial q$ are evaluated in BMATRIX. These partials are then numerically integrated to obtain $\partial r/\partial q$.

Using the acceleration partials as evaluated in BMATRIX, if

36-2 of 4

$$B = \left\{ \begin{array}{cccc} 0 & 0 & \dots & 0 \\ 0 & 0 & \dots & 0 \\ 0 & 0 & \dots & 0 \\ \frac{\partial \ddot{x}}{\partial q_1} & \frac{\partial \ddot{x}}{\partial q_2} & \dots & \frac{\partial \ddot{x}}{\partial q_n} \\ \frac{\partial \ddot{y}}{\partial q_1} & \frac{\partial \ddot{y}}{\partial q_2} & \dots & \frac{\partial \ddot{y}}{\partial q_n} \\ \frac{\partial \ddot{z}}{\partial q_1} & \frac{\partial \ddot{z}}{\partial q_2} & \dots & \frac{\partial \ddot{z}}{\partial q_n} \end{array} \right\}$$

$$U = \left\{ \begin{array}{ccc} \frac{\partial x}{\partial x_0} & \dots & \frac{\partial x}{\partial z_0} \\ \frac{\partial y}{\partial x_0} & \dots & \frac{\partial y}{\partial z_0} \\ \frac{\partial z}{\partial x_0} & \dots & \frac{\partial z}{\partial z_0} \\ \frac{\partial \dot{x}}{\partial x_0} & \dots & \frac{\partial \dot{x}}{\partial z_0} \\ \frac{\partial \dot{y}}{\partial x_0} & \dots & \frac{\partial \dot{y}}{\partial z_0} \\ \frac{\partial \dot{z}}{\partial x_0} & \dots & \frac{\partial \dot{z}}{\partial z_0} \end{array} \right\}$$

and

36-3 of 4

$$\Gamma = \left\{ \begin{array}{ccc} \frac{\partial x}{\partial q_1} & \dots & \frac{\partial x}{\partial q_n} \\ \frac{\partial y}{\partial q_1} & \dots & \frac{\partial y}{\partial q_n} \\ \frac{\partial z}{\partial q_1} & \dots & \frac{\partial z}{\partial q_n} \\ \frac{\partial \dot{x}}{\partial q_1} & \dots & \frac{\partial \dot{x}}{\partial q_n} \\ \frac{\partial \dot{y}}{\partial q_1} & \dots & \frac{\partial \dot{y}}{\partial q_n} \\ \frac{\partial \dot{z}}{\partial q_1} & \dots & \frac{\partial \dot{z}}{\partial q_n} \end{array} \right\}$$

then the partials $\partial r/\partial q$ are obtained from the following integral using Simpson's method:

$$\Gamma(t) = U(t) \int_{t_0}^{t_U^{-1}} (t^*)B(t^*) dt^*$$

The Simpson procedure uses an integration step based on the data times; if no data times exist in the interval, the time steps employed in the trajectory integration are used. The inverse of the U matrix is obtained not by the usual numerical methods but by an inspection method as defined in the subroutine INSPC.

USE

CALL GAMAT Result of the integration is placed on the combined ephemeris tape for use as needed.

CODING INFORMATION

36-4 of 4

Length of subroutine is 369 (10) or 561 (8) words.

REFERENCE

Anderson, John D., RFP 312-37, August 29, 1961.

IDENTIFICATION

37

GERTA/GREOF

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

GERTA writes the partials $\partial F/\partial Q$ (the "phi vector") on the MMP output tape, B7-SYSUT9. Each set of partials is tagged by time, receiver, transmitter, and data type.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 WRITEB/ENDFIL
 FLAT

USE

CALL GERTA

CALL GREOF for end-of-file

COMMON input:

CPPHI	partials $\partial F/\partial Q$ (1 x 20)
TOB	time, d. p. sec past 1950.0
IR	receiver ID
IX	transmitter ID
ID	data type ID

CODING INFORMATION

Length of subroutine is 141 (10) or 215 (8) words.

IDENTIFICATION

38-1 of 2

GHADP

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

GHADP calculates the Greenwich hour angle for the time given in COMMON location TOB. Intermediate calculations are in double precision. Output is in COMMON location GHA.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

(DFDP)

(DFAD)

(DFMP)

(DFSB)

DMOD

COS

METHOD

$$\alpha_G = 100^\circ.0755426 + 0^\circ.985647346d + 2^\circ.9015 \\ \times 10^{-13} d^2 + \omega s + \Delta\lambda \cos \bar{\epsilon} \pmod{360^\circ}$$

where α_G = Greenwich hour angle

d = integer days past 1950.0

s = sec past 0^h of dth day

ω = Earth rotation rate

$\bar{\epsilon}$ = mean obliquity

$\Delta\lambda$ = nutation in longitude

The mean obliquity is given by

$$\bar{\epsilon} = 23^\circ.445759 - 0^\circ.1309404T - 0^\circ.88 \times 10^{-6} T^2 + 0^\circ.5 \times 10^{-6} T^3$$

where T = Julian centuries past 1950.0

USE

38-2 of 2

CALL GHADP

COMMON input:

TOB observation time, d. p. sec past 1950.0

OMEGA Earth rotation rate, deg/sec

DLO nutation in longitude, deg

COMMON output:

GHA Greenwich hour angle, deg

CODING INFORMATION

Length of subroutine is 208 (10) or 320 (8) words.

REFERENCE

Holdridge, D. B., TR 32-223, March 2, 1962.

IDENTIFICATION

39

IMPAR

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

IMPAR computes the partials of impact time with respect to the estimated and considered parameters.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

FORM

METHOD

$$\frac{\partial T_I}{\partial Q} = \frac{-1}{\underline{\rho} \cdot \underline{\dot{\rho}}} \left(\underline{\rho} \cdot \frac{\partial \underline{\rho}}{\partial Q} \right)$$

where

T_I = impact time

Q = parameter

$\underline{\rho}$ = target centered probe position

$\underline{\dot{\rho}}$ = target centered probe velocity

USE

CALL IMPAR

PZE XTAR geocentric target position and velocity

COMMON input:

X geocentric probe position and velocity (1 × 6)

U partials $\partial \underline{\rho} / \partial \underline{r}_0$ (6 × 6)

GMAT partials $\partial \underline{\rho} / \partial q$ (6 × 11)

LLIST physical constants flags (11)

COMMON output:

QA partials $\partial T_I / \partial q$ (1 × 11)

QB partials $\partial T_I / \partial \underline{r}_0$ (1 × 6)

CODING INFORMATION

Length of subroutine is 127 (10) or 177 (8) words.

IDENTIFICATION

40

INSPC

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

INSPC inverts the U matrix of variational partials by inspection.

RESTRICTIONS

COMMON break: 47675

METHOD

If the U matrix is partitioned

$$U = \begin{bmatrix} U_{11} & | & U_{12} \\ \hline U_{21} & | & U_{22} \end{bmatrix}$$

then its inverse may be written as

$$U^{-1} = \begin{bmatrix} U_{22}^T & | & -U_{12}^T \\ \hline -U_{21}^T & | & U_{11}^T \end{bmatrix}$$

USE

CALL INSPC

COMMON input:

U U matrix of variational partials (6 x 6)

COMMON output:

UINV U⁻¹

CODING INFORMATION

Length of subroutine is 200 (10) or 310 (8) words.

REFERENCE

Anderson, John D., TM 312-409, March 24, 1964.

IDENTIFICATION

41

INTR1/BNTR2

Alan D. Rosenberg and Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

This version of INTR1 uses the latest version of the JPL Ephemeris routines, EPHEM, to obtain the necessary planetary information. INTR1 adjusts this output so it matches output from the original INTR1 rotated to true of date for the ODP.

RESTRICTIONS

- a. Portions of the ODP COMMON storage must be made available to the routine. ERROR return for planetary ephemeris tape error.
- b. Subroutines used:
 - EPHEM
 - PNUT
 - ROT
 - ERROR

USE

- | | | |
|------|-------|---|
| CALL | BNTR2 | Enter one time to set up blocks of storage necessary to EPHEM |
| CALL | INTR1 | Time is used from COMMON |

CODING INFORMATION

Length of subroutine is 2188 (10) or 4214 (8) words.

REFERENCES

- a. Holdridge, D. B., TR 32-223, Space Trajectories Program for the IBM 7090 Computer, March 2, 1962.
- b. Peabody, P. R., Scott, J. F., Orozco, E. G., TR 32-580, User's Description of JPL Ephemeris Tapes, March 2, 1964.

IDENTIFICATION

42

IXTAB

Michael R. Warner, JPL
 IBM 7094 Fap
 January 4, 1965

PURPOSE

IXTAB does a lookup in the table of transmitter ID's and frequencies by means of a time argument.

RESTRICTION

COMMON break: 47675

USE

CALL IXTAB
 PZE N

N is the location of a flag which is zero for the initialization entry and non-zero for the normal entry. The initialization entry returns the lookup pointer to the first point in the table.

COMMON input:

TOB lookup time, d.p. sec past 1950.0
 EPOCH ODP epoch, d.p. sec past 1950.0
 YY1 table of times, sec past epoch (1 × 90)
 YY1-90 table of transmitter ID's (1 × 90)
 YY1-180 table of frequencies (1 × 90)

COMMON output:

IX transmitter ID
 XFREQ transmitter frequency if IX ≠ 0
 TFREQ transponder frequency if IX ≠ 0

CODING INFORMATION

Length of subroutine is 54 (10) or 66 (8) words.

IDENTIFICATION

43-1 of 2

KINE

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

KINE reads the ODG data file or the ODP residual file from disk. Each entry loads COMMON with the contents of the next sequential logical record. KINE uses the REDE entry of the buffered disk routine DISCBU.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 - REDE
 - ERROR
- c. Error conditions:
 - Disk parity error
 - Attempted read outside of file limits

USE

CALL	KINE				
PZE	RVECO	} for ODG data reading	PZE	RVEC	} for ODP residual reading
PZE	RVECW		PZE	RVECD	
PZE	= 1001		PZE	= 1	
PZE	= 2000		PZE	= 1000	
PZE	IF6	1B17 = EOF, 2B17 = normal			

COMMON output:

TOB	observation time, d. p. sec past 1950.0
RVEC	array of computed data types
RVECO	array of observed data types
RVECW	array of weight code-words
RVECD	array of residual-weight words
PASS	pass identification
XFREQ	transmitter frequency, cps
TFREQ	transponder frequency, cps
IR	receiver identification
IX	transmitter identification
TAU	doppler averaging time, sec

The ODG logical record format is as follows:

43-2 of 2

<u>Word</u>	0	word count (2N + 5, B17) or -0 for EOF
	1	TOB
	2	TOB - 1
	3	XFREQ if IX \neq 0 TFREQ if IX \neq 0
	4	PASS
	5	<u>bit</u> S 1 if L - S band 0 if L or S band
		1 1 if RVECO \neq 0
		2 1 if RVECO - 1 \neq 0
		3 1 if RVECO - 2 \neq 0
		.
		.
		.
	11	1
	12-15	not used
	16	1 if L - S or S band 0 if L band
	17	1 if optical data
	18-21	IX
	22-25	IR
	26-35	TAU.
	6	first member of RVECO which is non-zero
	7	<u>bits</u> S-17 weight codeword corresponding to word 6 18-35 data sample rate, sec
	8	second member of RVECO which is non-zero
	9	second weight-sample rate word

2N + 4 Nth member of RVECO which is non-zero

2N + 5 Nth weight-sample rate word

The corresponding residual record format is found in the documentation of subroutine FILL.

CODING INFORMATION

Length of subroutine is 142 (10) or 216 (8) words.

IDENTIFICATION

44

LOCO

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

This is the control routine for the link which restores the ODP COMMON from disk following computation of the trajectory. In addition it controls preparation of the combined ephemeris on tape A4.

RESTRICTIONS

- a. ERROR condition: disk error indicated by DCP
- b. COMMON break: 47675
- c. Subroutines used:

KOOL

TAPIO

INTR1

EPPHEM

FLAK

TYPRYT

USE

This is one of the basic links of the ODP in that it restores COMMON and prepares the ephemeris. It is called under control of the JPTRAJ monitor.

CODING INFORMATION

Length of subroutine is 116 (10) or 164 (8) words.

IDENTIFICATION

45-1 of 2

LOOKUP/KOOL

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

To perform look-up on the probe ephemeris generated by SPACE.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

 ERROR Called when time incompatible with the ephemeris which has been generated.

 REDE

METHOD

The subroutine computes the coefficients for a 5th degree Lagrangian interpolation and provides values for the position, velocity, acceleration and jerk (3rd derivative) of the probe, variational equations, nutation in longitude and nutation in obliquity. (Flow chart is included)

$$y(x) = \sum_{k=0}^w l_k(x) f(x_k)$$

where

$$l_i(x) = \frac{\pi(x)}{(x - x_1) \pi'(x_1)}$$

$$= \frac{(x - x_0) \cdots (x - x_{i-1})(x - x_{i+1}) \cdots (x - x_n)}{(x_i - x_0) \cdots (x_i - x_{i-1})(x_i - x_{i+1}) \cdots (x_i - x_n)}$$

For each time that LOOKUP is called the following information is stored in COMMON

X	}	Probe vector
Y		
Z		
XDOT		
YDOT		
ZDOT	}	(36) Variational equations
U		
$\partial x/\partial x_0, \partial x/\partial y_0, \partial x/\partial z_0, \partial x/\partial \dot{x}_0 \dots \partial z/\partial x_0 \dots \partial z/\partial \dot{z}_0$		
DLO		Nutation in longitude
DOB		Nutation in obliquity
XAC	}	Probe acceleration
YAC		
ZAC		
XJERK	}	Probe jerk (3rd derivative)
YJERK		
ZJERK		

USE

CALL LOOKUP (n) n = number of items to be interpolated, 6 or 49. When 6, the probe vector only is given.

CALL KOOL to provide a logical reset on the probe ephemeris file and to set flag for LOOKUP to read 3 physical records needed on first entry to the subroutine.

CODING INFORMATION

Length of subroutine is 314 (10) or 472 (8) words.

REFERENCE

Hildebrand, F. B., Introduction to Numerical Analysis, McGraw-Hill, New York, 1956.

IDENTIFICATION

46

MAMUL

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To multiply two matrices and store in a third array.

RESTRICTION

Maximum array: 20 x 20

USE

CALL MAMUL (A, B, C, M, N, L)
A(M, L) * B(L, N) = C (M, N)

CODING INFORMATION

Length of subroutine is 87 (10) or 127 (8) words.

IDENTIFICATION

47

MAPOUT

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

MAPOUT reads all matrices which were computed and stored on disk in Link 2. Here they are printed and in some cases punched. Also, the mapping information is saved for the MMP (MATRIX MANIPULATOR PROGRAM) input tape.

RESTRICTIONS

a. ERROR condition: disk error indicated by DCP

b. COMMON break: 47055

c. Subroutines used:

PRIM

FLAK

REFORM

PUMA

PRAM

DCP

IOCS

TYPRYT

OFFSYS

TACCOM

USE

This link is called under control of the JPTRAJ Source Deck.

CODING INFORMATION

Length of subroutine is 866 (10) or 1542 (8) words.

IDENTIFICATION

48

MAXIM

Melba Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

Checks data tape, mapping times, predictions, pointing times, etc. so that the trajectory can be run a minimum length of time and still satisfy all demands on the probe ephemeris.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

KINE

MOCT

QUIZ

(DFAD)

(DFSB)

METHOD

Flow chart is included.

USE

CALL MAXIM

CODING INFORMATION

Length of subroutine is 315 (10) or 473 (8) words.

IDENTIFICATION

49

MOCT

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

MOCT obtains the latest occultation or impact time for determining the trajectory link time stop.

RESTRICTION

COMMON break: 47675

USE

CALL MOCT

PZE LAST latest occultation/impact time, d. p. sec past 1950.0

COMMON input:

PTFD occultation/impact input area

DXDR-27 occultation/impact flag

CODING INFORMATION

Length of subroutine is 71 (10) or 107 (8) words.

IDENTIFICATION

50

ND2F

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

This is a SHARE routine which has been modified so that it is possible to start at a pre-determined portion of the random number generator.

RESTRICTION

COMMON break: 47675

METHOD

RANO, an octal number, can be input into the ODP data.

This number is chosen from a pre-calculated set.

USE

To provide noise on a simulated data tape.

CODING INFORMATION

Length of subroutine is 39 (10) or 47 (8) words.

IDENTIFICATION

51

NOMNL

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

NOMNL stores the nominal values of all constants, other than the weighting tables and the target-dependent solar pressure constants.

RESTRICTION

COMMON break: 47675

USE

CALL NOMNL

This subroutine outputs into approximately 150 COMMON locations.

CODING INFORMATION

Length of subroutine is 499 (10) or 1051 (8) words.

REFERENCES

- a. Clarke, Victor C., TR 32-604, March 6, 1964.
- b. Scott, James F., IOM 317.21/318, September 1, 1964.

IDENTIFICATION

52

NORMAY

Melba Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To normalize a matrix on its diagonal terms.

RESTRICTIONS

- a. Maximum matrix: 20×20
- b. Subroutines used:

SQRT

USE

CALL NORMAY (XMAT, YMAT, N)

XMAT = matrix A to be normalized

YMAT = location to store normalized matrix

N = order of matrix

CODING INFORMATION

Length of subroutine is 169 (10) or 251 (8) words.

IDENTIFICATION

53

NOUT

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

NOUT prints comments indicating the rows and columns of the normal matrix which were deleted by subroutine STPREG during inversion:

VARIABLE NO. XX REJECTED BY DIAGONAL TEST.

RESTRICTION

Subroutines used:

PROUT

USE

CALL NOUT

PZE INOUT

INOUT is a 1×20 array in which STPREG has indicated the status of each variable.

COMMON input:

PRIFIL PROUT file control block for printing.

CODING INFORMATION

Length of subroutine is 64 (10) or 100 (8) words.

IDENTIFICATION

54-1 of 3

OBTOX

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

OBTOX computes the partials of the ODP data types with respect to the probe position and velocity at the observation time.

RESTRICTION

COMMON break: 47675

METHOD

$$\begin{array}{l}
 \left. \begin{array}{l}
 \frac{\partial \rho_i}{\partial \underline{x}} = \underline{L} \\
 \frac{\partial \rho_i}{\partial \underline{\dot{x}}} = 0
 \end{array} \right\} \text{(slant range partials)} \\
 \\
 \left. \begin{array}{l}
 \frac{\partial \dot{\rho}_i}{\partial x} = \frac{1}{\rho_i} (\dot{x} + \omega y_1 - \dot{\rho}_1 L_x) \\
 \frac{\partial \dot{\rho}_i}{\partial y} = \frac{1}{\rho_i} (\dot{y} - \omega x_1 - \dot{\rho}_1 L_y) \\
 \frac{\partial \dot{\rho}_i}{\partial z} = \frac{1}{\rho_i} (\dot{z} - \dot{\rho}_1 L_z) \\
 \frac{\partial \dot{\rho}_i}{\partial \underline{\dot{x}}} = \underline{L}
 \end{array} \right\} \text{(range rate partials)} \\
 \\
 \left. \begin{array}{l}
 \frac{\partial \gamma_i}{\partial \underline{x}} = \frac{\underline{D}}{\rho_i} \\
 \frac{\partial \gamma_i}{\partial \underline{\dot{x}}} = 0
 \end{array} \right\} \text{(elevation angle partials)}
 \end{array}$$

$$\left. \begin{aligned} \frac{\partial \sigma_i}{\partial \underline{r}} &= \frac{\tilde{A}}{\rho_i} \\ \frac{\partial \sigma_i}{\partial \underline{\dot{r}}} &= 0 \end{aligned} \right\} \text{(azimuth angle partials)}$$

$$\left. \begin{aligned} \frac{\partial \delta_i}{\partial x} &= \frac{-\cos a_{ri} \sin \delta_i}{\rho_i} \\ \frac{\partial \delta_i}{\partial y} &= \frac{-\sin a_{ri} \sin \delta_i}{\rho_i} \\ \frac{\partial \delta_i}{\partial z} &= \frac{\cos \delta_i}{\rho_i} \\ \frac{\partial \delta_i}{\partial \underline{\dot{r}}} &= 0 \end{aligned} \right\} \text{(declination partials)}$$

$$\left. \begin{aligned} \frac{\partial a_i}{\partial x} &= \frac{\sin a_{ri}}{\rho_i \cos \delta_i} \\ \frac{\partial a_i}{\partial y} &= \frac{-\cos a_{ri}}{\rho_i \cos \delta_i} \\ \frac{\partial a_i}{\partial z} &= 0 \\ \frac{\partial a_i}{\partial \underline{\dot{r}}} &= 0 \end{aligned} \right\} \text{(hour angle partials)}$$

$$\frac{\partial f_{\underline{h}}}{\partial \underline{r}} = \frac{\Omega_2}{c} \frac{\partial \dot{\rho}_i}{\partial \underline{r}} \quad \underline{r} \rightarrow \underline{\dot{r}} \quad \text{(one-way doppler partials)}$$

$$\frac{\partial f_{c3i,q}}{\partial \underline{r}} = \frac{\Omega_4}{c} \left(\frac{\partial \dot{\rho}_i}{\partial \underline{r}} + \frac{\partial \dot{\rho}_q}{\partial \underline{r}} \right) \quad \underline{r} \rightarrow \underline{\dot{r}} \quad \text{(two-way doppler partials)}$$

$$\frac{\partial f_{3i,q}}{\partial \underline{r}} = \frac{\Omega_6}{c} \left(\frac{\partial \dot{\rho}_i}{\partial \underline{r}} + \frac{\partial \dot{\rho}_q}{\partial \underline{r}} \right) \quad \underline{r} \rightarrow \dot{\underline{r}} \quad (\text{three-way doppler partials})$$

$$\frac{\partial f_{d1i,j}}{\partial \underline{r}} = \frac{\Omega_2}{c} \left(\frac{\partial \dot{\rho}_i}{\partial \underline{r}} - \frac{\partial \dot{\rho}_j}{\partial \underline{r}} \right) \quad \underline{r} \rightarrow \dot{\underline{r}} \quad (\text{differenced one-way partials})$$

(all symbols used here are defined in the subroutine CATS documentation)

USE

CALL OBTOX

COMMON input:

ELX \underline{L}
 TAX $\underline{\tilde{A}}$
 TDX $\underline{\tilde{D}}$
 RVEC ODP computed data type array
 X $\underline{r}, \dot{\underline{r}}$
 OMEGA ω , deg/sec
 RAI α_{ri} , deg
 CZ2 Ω_2
 CZ4 Ω_4
 CZ6 Ω_6
 VELC c , km/sec

COMMON output:

POBX $\partial F / \partial \underline{r}, \partial F / \partial \dot{\underline{r}}$

CODING INFORMATION

Length of subroutine is 189 (10) or 275 (8) words.

REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Anderson, John D., TM 312-409, March 24, 1964.

IDENTIFICATION

55

OCIM

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

OCIM reads the ODP combined ephemeris tape, extracting occultation and impact time parameters necessary for partials calculation.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines called:

IMPAR

OCPAR

READB

FLAT

ERROR

c. Error conditions:

Ephemeris tape permanent redundancy

USE

CALL OCIM

COMMON input:

DXDR-6	target (1 = Mars 2 = Moon 3 = Venus)
DXDR-27	1B17 = occultation time 2B17 = impact time
XMOO	geocentric Moon vector
XVEN	geocentric Venus vector
XMAR	geocentric Mars vector
PTFD	occultation-impact times input array
X	geocentric probe vector
XAC	geocentric probe accelerations
XJERK	geocentric probe jerks
RVECD	residual
RVECW	weight
DINT	physical constants partials
T	corrected occultation/impact time

IDENTIFICATION

56-1 of 3

OCPAR

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

OCPAR computes the partials of occultation time with respect to the estimated and considered parameters.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

FORM

METHOD

$$\frac{\partial T_0}{\partial Q} = \frac{\rho \cdot \rho_m \left(\frac{\rho_m \cdot \frac{\partial \rho_m}{\partial Q}}{\rho_m^2} + \frac{\rho \cdot \frac{\partial \rho}{\partial Q}}{\rho^2} \right) - \rho \cdot \frac{\partial \rho_m}{\partial Q} - \rho_m \cdot \frac{\partial \rho}{\partial Q} + \frac{|\rho| |\rho_m| \sin \phi_p R_m}{\rho_m^2 \sqrt{\rho_m^2 - R_m^2}} \rho_m \cdot \frac{\partial \rho_m}{\partial Q}}{\rho \cdot \rho_m \left(\frac{\rho_m \cdot \dot{\rho}_m}{\rho_m^2} + \frac{\rho \cdot \dot{\rho}}{\rho^2} \right) - \rho \cdot \rho_m - \rho_m \cdot \rho + \frac{|\rho| |\rho_m| \sin \phi_p R_m}{\rho_m^2 \sqrt{\rho_m^2 - R_m^2}} (\rho_m \cdot \rho_m)}$$

where

$$\rho_m = r_m - R_i$$

R_m = radius of target

r_m = geocentric target vector

$$\phi_p = \cos^{-1} \frac{\underline{\rho} \cdot \underline{\rho}_m}{|\underline{\rho}| |\underline{\rho}_m|}$$

$\underline{\rho}$ = topocentric probe vector

\underline{R}_i = geocentric station vector

USE

CALL OCPAR

PZE XTAR \underline{r}_m , km

PZE AA $A = \underline{\Gamma} \cdot \underline{\dot{\Gamma}}$

PZE AB $B = \underline{\beta} \cdot \underline{\dot{\beta}}$

PZE AC $C = \sqrt{\Gamma^2 \beta^2 - (\underline{\Gamma} \cdot \underline{\beta})^2}$

PZE AK $K = \underline{\Gamma} \cdot \underline{\dot{\beta}} + \underline{\beta} \cdot \underline{\dot{\Gamma}}$

PZE AM $M = R_m C / \beta^2 \sqrt{\beta^2 - R_m^2}$

PZE GDB $\underline{\Gamma} \cdot \underline{\beta}$

PZE BETA $|\underline{\beta}|$ ($\underline{\beta} = \underline{\rho}_m$)

PZE GAMMA $|\underline{\Gamma}|$ ($\underline{\Gamma} = \underline{r} - \underline{R}_i$)

PZE XMX $\underline{\rho}_m \cdot \underline{r}$

PZE XXI $\underline{r} \cdot \underline{R}_i$

PZE SI \underline{R}_i

COMMON input:

U partials $\partial \rho / \partial \underline{r}_0$ (6×6)

GMAT partials $\partial \underline{\rho} / \partial q$ (6×11)

LLIST physical constants flags (11)

RI $|\underline{R}_i|$, km

NLIST station locations flags (3×15)

COSRAI $\cos a_{ri}$

SINRAI $\sin a_{ri}$

COSPHI $\sin \phi_i$

COMMON output:

56-3 of 3

QA partials $\partial T_0 / \partial q$
QB partials $\partial T_0 / \partial \underline{r}_0$
DIM3 partials $\partial T_0 / \partial s_i$

CODING INFORMATION

Length of subroutine is 333 (10) or 515 (8) words.

REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Liu, Anthony, RFP 312-136, May 20, 1963.

IDENTIFICATION

57-1 of 2

ODATA/ONLIN/.....
 Michael R. Warner, JPL
 IBM 7094 Fap
 January 4, 1965

PURPOSE

ODATA performs the following functions:

- a. In conjunction with the conversion subroutine CARDS, reads ODP control cards and the symbolic and/or numeric data cards. The cards may be read from the on-line reader or input tape A2 (mode 4) or the user area reader (mode 2).
- b. Manipulates the symbolic and numeric data and stores them in the appropriate COMMON locations for subsequent ODP use.
- c. Under sense switch (or console key) control, terminates ODP execution.
- d. Prints the trajectory target on the administrative (or on-line) printer.
- e. Stores the target-dependent solar pressure constants.
- f. On option, reads the previously computed injection conditions, constants, and covariance matrix from disk.
- g. Applies Gauss' constraint to the planetary masses:

$$GM_s = 3.9640160 \times 10^{-14} a_e^3$$

$$GM_v = \frac{M_v}{M_s} GM_s$$

$$GM_r = \frac{M_r}{M_s} GM_s$$

$$GM_j = \frac{M_j}{M_s} GM_s$$

where

a_e = astronomical unit

s, v, r, j = subscripts denoting the Sun, Venus, Mars, Jupiter.

- h. On option, applies the scaling constraint to the Earth radius:

$$R_{em} = 86.315745 (GM_e + GM_m)^{1/3}$$

where

57-2 of 2

R_{em} = earth radius for scaling the lunar ephemeris.

RESTRICTIONS

- a. If any of the following error conditions are detected, ODATA calls subroutine ERROR for the appropriate comment printout and subsequent action:
 - 1. Error in symbolic or numeric input.
 - 2. Checksum or redundancy error.
 - 3. Cards not received within time limit (mode 2).
 - 4. Illegal card in deck (mode 2)
 - 5. End-of-file indication (mode 4)
 - 6. Data on input card inconsistent with ODP requirements.
 - 7. Error in reading disk.
- b. COMMON break: 47675
- c. Subroutines called:

TYPRYT
 IOCS - DEFINE/ATTACH/OPEN/CLOSE
 CARDS
 OFFSYS
 ERROR
 FINSYS
 READS/MOOPH1/ODOFF
 FLOT
 SPHX
 STPREG
 DIAG
 DISCBU - REDE
 EXP(3)

USE

CALL ODATA

ODATA uses most of the ODP COMMON map. Section IVB has the COMMON listing. The input descriptions are in Section IVA.

ONLIN is used for returning to ODATA after a corrected card input error (TTR \$ONLIN).

..... represents the entry point for the JPTRAJ program control block of LA9.

CODING INFORMATION

Length of subroutine is 1723 (10) or 3273 (8) words.

IDENTIFICATION

58

ORBEQ

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

ORBEQ rotates the encounter noise moment matrix from the ODP R, T plane to the midcourse program R, T plane.

RESTRICTIONS

a. COMMON break: 47055

b. Subroutines used

ARTAN

COS

SIN

METHOD

$$\theta = \tan^{-1} \frac{\underline{B} \cdot \underline{R}_{m/c}}{\underline{B} \cdot \underline{T}_{m/c}} - \tan^{-1} \frac{\underline{B} \cdot \underline{R}_{odp}}{\underline{B} \cdot \underline{T}_{odp}}$$

$$R = \begin{bmatrix} \cos \theta & -\sin \theta & 0 \\ \sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

USE

CALL ORBEQ
 PZE BRO ODP B · R, km
 PZE BTO ODP B · T, km
 PZE BRM M/C B · T, km
 PZE BTM M/C B · T, km
 PZE CNO ODP noise moment matrix (6 × 6)
 PZE CNM M/C noise moment matrix (3 × 3)

CODING INFORMATION

Length of subroutine is 119 (10) or 167 (8) words.

IDENTIFICATION

59

OZ

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

OZ is the control program for the calculation of the closest approach parameters.

RESTRICTIONS

a. COMMON break: 47055

b. Subroutines used:

UMAT

TYPRYT

IOCS

OFFSYS

USE

When impact or closest approach parameters are requested, this link is called under control of the JPTRAJ Source Deck.

CODING INFORMATION

Length of subroutine is 766 (10) or 1376 (8) words.

IDENTIFICATION

60-1 of 2

PARAM

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

PARAM outputs the encounter parameters page of the ODP printout. It also loads the buffers for COMENT output.

RESTRICTIONS

- a. COMMON break: 47055
- b. Subroutines called:

SQRT

FIXT

PROUT

TNORM

ORBEQ

USE

CALL	PARAM	
PZE	CNO	noise moment matrix of encounter parameters (6 × 6)
PZE	BRO	$\underline{B} \cdot \underline{R}$, km (ODP plane)
PZE	BTO	$\underline{B} \cdot \underline{T}$, km (ODP plane)
PZE	ELPS	miss ellipse configuration (1 × 3 - semi-major axis, semi-minor axis, orientation angle)
PZE	A	semi-major axis of encounter conic
PZE	UNJ	$\partial M / \partial Q_0$ matrix of encounter partials
PZE	TL	linearized time of flight, days
PZE	UT	$\partial M / \partial Q$ matrix of encounter partials

COMMON input:

DXDR-6	target identification
DXDR-21	$\underline{B} \cdot \underline{R}$ (midcourse plane)
DXDR-20	$\underline{B} \cdot \underline{T}$ (midcourse plane)
DXDR-2	COMENT update flag
GRAVE	GM_{Earth} , km ³ /sec ²
GRAVR	GM_{Mars}
GRAVV	GM_{Venus}

GRAVM	GM_{Moon}	60-2 of 2
GRAVS	GM_{Sun}	
TFIN	time of impact/closest approach, d.p. seconds past 1950.0	
COMTS	ODP page heading, BCD	
PRIFIL	PROUT file control block	
XNJ	ODP initial conditions (1 × 6), d.p.	
EPOCH	ODP epoch, d.p. sec past 1950.0	
XDOT	probe velocity at encounter, km/sec (1 × 3)	

COMMON output:

COMBU	midcourse program COMMENT parameters
COMBW	trajectory program COMMENT parameters

CODING INFORMATION

Length of subroutine is 720 (10) or 1320 (8) words.

IDENTIFICATION

61-1 of 2

PARLEY
 Melba W. Nead, JPL
 IBM 7094 Fap
 January 4, 1965

PURPOSE

To write time, partials, position and velocity of probe and planets, etc., on tape for each data point while the data is being processed.

RESTRICTIONS

- a. COMMON break: 47675
- b. Cannot be used in SFOF MODE 2 (real-time mode).
- c. Subroutines used:

FLAT
 TAPIO

USE

CALL PARLEY

The following items are written on a scratch tape to be read and printed in a subsequent link:

TOB		Observation time
GHA		Greenwich Hour Angle
X	}	Probe vector
Y		
Z		
XDOT		
YDOT		
ZDOT		
U	(36)	Variational equations
DLO		Nutation in longitude
DOB		Nutation in obliquity
XAC	}	Probe Acceleration
YAC		
ZAC		
XJERK	}	Probe jerk (3rd derivative)
YJERK		
ZJERK		
XSUN	(6)	Sun vector

XVEN	(6)	Venus vector	} XTARG vector only	61-2 of 2
XMOO	(6)	Moon vector		
XMAR	(6)	Mars vector		
CPPHI	(20)	CAP PHI matrix of partials		
CPTHT	(20)	CAP THETA matrix of partials		
DELT		Light time correction		
DELT2		Second light time correction		
DHA		} Refraction corrections		
DDEC				
DAZ				
DEL				
DRDOT				
DRVEC				

CODING INFORMATION

Length of subroutine is 30 (10) or 36 (8) words.

REFERENCE

Null, George, W., RFP 312-179, August 15, 1963.

IDENTIFICATION

62

PARSH

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

This link is called only when the COMENT region is to be updated, or when in MODE 4, the printing of the partials is requested. The program is called under control of the JPTRAJ monitor.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutine used:

COMOUT

WASH

FGDOUT

DISCBU

USE

Control program for writing COMENT on disk or printing partials.

CODING INFORMATION

Length of subroutine is 662 (10) or 1226 (8) words.

IDENTIFICATION

63

PERNOD/WOLF

Charles Coltharp/Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

PERNOD accumulates statistics on data residuals. It is called during the processing of each residual. WOLF is called at the end-of-file for each station's residuals. It obtains the standard deviation, root-mean-square, mean, and second moment for each data type and each pass identification.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

FIXT

PROUT/FGDOUT

SQRT

COLA

USE

CALL PERNOD after each residual is read

CALL WOLF after each station is completed

COMMON input:

TOB observation time, d. p. seconds past 1950.0

RVECD residual array (1 × 11)

ITNO iteration number

PASS pass identification

IR receives identification

PRIFIL IOCS printing file control block

CODING INFORMATION

Length of subroutine is 323 (10) or 503 (8) words.

IDENTIFICATION

64

PM360
 Michael R. Warner, JPL
 Fortran II, Version 3
 January 4, 1965

PURPOSE

PM360 adjusts angle residuals by 360 deg if they are greater than 180 deg or less than -180 deg.

RESTRICTIONS

COMMON break: 47675

USE

CALL PM360

COMMON input:

DECD	declination residual
HAD	hour angle residual
AZID	azimuth residual

COMMON output:

DECD	} residuals adjusted by 360 deg if necessary
HAD	
AZID	

CODING INFORMATION

Length of subroutine is 40 (10) or 50 (8) words.

IDENTIFICATION

65-1 of 2

PNUT

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

PNUT calculates the rotation matrix for nutation-precession.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

SIN

COS

METHOD

$$N = \begin{bmatrix} 1 & -\Delta\lambda \cos \bar{\epsilon} & -\Delta\lambda \sin \bar{\epsilon} \\ \Delta\lambda \cos \bar{\epsilon} & 1 & -\Delta\epsilon \\ \Delta\lambda \sin \bar{\epsilon} & \Delta\epsilon & 1 \end{bmatrix}$$

$$a_{11} = 1 - 0.29697 \times 10^{-3} T^2 - 0.13 \times 10^{-6} T^3$$

$$a_{12} = -0.02234988T - 0.676 \times 10^{-5} T^2 + 0.221 \times 10^{-5} T^3$$

$$a_{13} = -0.00971711T + 0.207 \times 10^{-5} T^2 + 0.96 \times 10^{-6} T^3$$

$$a_{21} = -a_{12}$$

$$a_{22} = 1 - 0.24976 \times 10^{-3} T^2 - 0.15 \times 10^{-6} T^3$$

$$a_{23} = -0.10859 \times 10^{-3} T^2 - 0.3 \times 10^{-7} T^3$$

$$a_{31} = -a_{13}$$

$$a_{32} = a_{23}$$

65-2 of 2

$$a_{33} = 1 - 0.4721 \times 10^{-4} T^2 + 0.2 \times 10^{-7} T^3$$

where

$\Delta\lambda$ = nutation in longitude

$\Delta\epsilon$ = nutation in obliquity

$\bar{\epsilon}$ = mean obliquity (see writeup for GHADP)

T = Julian centuries past 1950.0

USE

CALL PNUT

COMMON input:

T time, d.p. sec past 1950.0

DOB nutation in obliquity, deg

DLO nutation in longitude, deg

COMMON output:

ROTMX nutation-precession matrix (NA)

CODING INFORMATION

Length of subroutine is 222 (10) or 336 (9) words.

REFERENCE

Holdridge, D. B., TR 32-223, March 2, 1962.

IDENTIFICATION

66

POINT

Melba W. Nead, JPL
Fortran II, Version 3
January 4, 1965

PURPOSE

To generate the pointing predictions for the tracking stations.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

- COEF
- FILL
- OFFSYS
- CORR
- GHADP
- REAP
- DOPLR
- IXTAB
- (DFAD)

USE

CALL POINT

CODING INFORMATION

Length of subroutine is 196 (10) or 304 (8) words.

REFERENCE

Trask, D.W., RFP 312-37, Addendum 6, April 4, 1962.

IDENTIFICATION

67

PONT

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To compute the times at which the combined ephemeris shall be written for a simulated data tape or predictions.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

GAMAT

INTR1

LOOKUP

OFFSYS

RITEM

TIMER

(DFAD)

Flow chart included

USE

CALL PONT

CODING INFORMATION

Length of subroutine is 201 (10) or 311 (8) words.

IDENTIFICATION

68

PRAMS

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

To save the U matrix when it is calculated by COMIMP. Core space forces the calculation and the print-out of these matrices to be done in separate links.

RESTRICTIONS

a. ERROR condition: disk error indicated by DCP

b. Subroutines used:

FLAK

DCP

USE

CALL PRAMS

CODING INFORMATION

Length of subroutine is 18 (10) or 22 (8) words. —

IDENTIFICATION

69

PREDA

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

This is the main program of the link which computes predictions for the DSIF stations or prepares a simulated data tape.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

POINT
DATAPE
TYPRYT
TAPIO

USE

This link is called under control of the JPTRAJ Source Deck when predictions or a data tape are requested.

CODING INFORMATION

Length of subroutine is 39 (10) or 47 (8) words.

IDENTIFICATION

70-1 of 2

PRIM/PRAM

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

To print any matrix up to 20×20 from a possible list of 63. Included is the capability of punching the array in a format acceptable as input back into the ODP. When step-mapping, probe position and velocity and the mapping matrix are automatically punched.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

BIBCD
 FLAPR
 PRIME
 NORMAY
 PRIT
 DIAGO
 FGDOU
 PROUT

USE

CALL PRIM
 PZE , , N

where N is as follows:

1. Input J matrix of estimated parameters
2. Input covariance matrix of considered parameters
3. J inverse
4. Covariance matrix of estimated parameters
5. Correlation matrix of estimated parameters
6. Covariance matrix at impact
7. Covariance matrix mapped forward
8. J matrix
9. Input covariance matrix of estimated parameters
10. Correlations based on J matrix
11. U product matrix in step-mapping

CALL PRAM
TSX CAPU

70-2 of 2

Prints U matrix for mapping forward

To punch a matrix PUNCH=n under OTHER PARAMETER VALUES in the data input. This will punch the matrix only one time.

CODING INFORMATION

Length of subroutine is 787 (10) or 1423 (8) words.

REFERENCE

Trask, D.W., REP 312-37, Addendum 6, April 4, 1962.

IDENTIFICATION

71

PRIME/REFORM

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

To write impact times, probe and target vectors and the mapping matrices on the MATRIX MANIPULATOR PROGRAM input tape.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

TAPIO

FLAT

USE

CALL REFORM To read mapping times from preliminary storage tape. TFIN is added to those originally saved and all are written on the final MMP tape.

CALL PRIME To write the mapping matrices on the MMP tape.____

CODING INFORMATION

Length of subroutine is 98 (10) or 142 (8) words.

REFERENCE

Peterson, G. E., ED 218, May 15, 1964.

IDENTIFICATION

72

PRINQ/PRIT/APRIOR

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

To print DQ's, new Q's, old Q's, standard deviation of covariance matrix of estimated parameters, etc.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

FGDOUT
DIAG
CLOCK
PROUT
FLAPR
FIXT
BIBCD

USE

CALL PRINQ	Prints DQ's etc. of estimated parameter.
CALL PRIT	Entered with double precision time in seconds in ACC and MQ. These are converted to year, month, day, hour, minute, second, and stored in COMMON.
APRIOR	Buffer in which the standard deviation of the aprior covariance matrix is stored through the "WANT" capability of the JPTRAJ monitor.

CODING INFORMATION

Length of subroutine is 259 (10) or 403 (8) words.

REFERENCE

Trask, D.W., RFP 312-37, Addendum 6, April 4, 1962.

IDENTIFICATION

73

PRINT

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

PRINT controls the printing of the DQ's and the matrices which have been computed in the preceding link. Also, when occultation calculations are requested it reads this information from disk and prepares it for printing.

RESTRICTIONS

- a. ERROR condition: disk error indicated by DCP
- b. COMMON break: 47675
- c. Subroutines used:

- PRINQ
- PRIM
- FIXT
- PROUT
- FGDOUT
- FLAK
- DISCBU
- IOCS
- ENDIT
- OFFSYS

USE

Upon completion of the calculation of the data fitting link, the print link is called under control of the JPTRAJ MONITOR.

CODING INFORMATION

Length of subroutine is 1352 (10) or 2510 (8) words.

IDENTIFICATION

74

PROM

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To multiply two square matrices and store in a third array.

RESTRICTION

Maximum array: 20×20

USE

CALL PROM (A, B, C, N)

$A(N, N) * B(N, N) = C(N, N)$

CODING INFORMATION

Length of subroutine is 110 (10) or 156 (8) words.

IDENTIFICATION

75

QUEST

Melba W. Nead, JPL
Fortran II, Version 3
January 4, 1965

PURPOSE

Checks to ascertain that the times specified are greater than EPOCH plus light time correction. Exits with an error return if time is not compatible.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 - ERROR
 - SQRT
 - EXIT

USE

CALL QUEST

CODING INFORMATION

Length of subroutine is 117 (10) or 165 (8) words.

IDENTIFICATION

76

QUIZ

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

QUIZ allows a Fortran program to interrogate the COMMON location KEYS, which contains the "key settings" specified by the input deck.

RESTRICTION

COMMON break: 47675

USE

In Fortran,

B A = (octal representation of desired key setting)

IF (QUIZF (A)) S₁, S₂, S₁

where

S₁ is the normal return indicating that the key is set

S₂ is the normal return indicating that the key is not set

COMMON input:

KEYS key setting word

CODING INFORMATION

Length of subroutine is 2 (10) or 2 (8) words.

IDENTIFICATION

77-1 of 4

RATES

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

RATES obtains the topocentric slant range and its first three derivatives. Both the up-leg and down-leg are calculated using the light-time correction supplied by subroutine DOPLR. Four other range-dependent quantities are also obtained.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines called:

COS

SIN

SQRT

ARSIN

EXP(3

METHOD

$$a_{ri} = \lambda_i + a_G - 2\omega\Delta t \quad (\text{station right ascension})$$

$$X_i = R_i \cos \phi_i \cos a_{ri}$$

$$Y_i = R_i \cos \phi_i \sin a_{ri}$$

$$Z_i = R_i \sin \phi_i$$

$$\dot{X}_i = -\omega Y_i$$

$$\dot{Y}_i = \omega X_i$$

$$\ddot{X}_i = -\omega Y_i$$

$$\ddot{Y}_i = \omega X_i$$

(station coordinates and derivatives)

$$\begin{aligned} \ddot{X}_i &= -\omega Y_i \\ \ddot{Y}_i &= \omega X_i \end{aligned} \left. \vphantom{\begin{aligned} \ddot{X}_i &= -\omega Y_i \\ \ddot{Y}_i &= \omega X_i \end{aligned}} \right\} \begin{array}{l} \text{(station coordinates} \\ \text{and derivatives)} \end{array}$$

$$\begin{aligned} \underline{\rho}_i &= \underline{r} - \underline{R}_i \\ \dot{\underline{\rho}}_i &= \dot{\underline{r}} - \dot{\underline{R}}_i \\ \ddot{\underline{\rho}}_i &= \ddot{\underline{r}} - \ddot{\underline{R}}_i \\ \ddot{\underline{\rho}}_i &= \ddot{\underline{r}} - \ddot{\underline{R}}_i \end{aligned} \left. \vphantom{\begin{aligned} \underline{\rho}_i &= \underline{r} - \underline{R}_i \\ \dot{\underline{\rho}}_i &= \dot{\underline{r}} - \dot{\underline{R}}_i \\ \ddot{\underline{\rho}}_i &= \ddot{\underline{r}} - \ddot{\underline{R}}_i \\ \ddot{\underline{\rho}}_i &= \ddot{\underline{r}} - \ddot{\underline{R}}_i \end{aligned}} \right\} \begin{array}{l} \text{(topocentric coordinates)} \end{array}$$

$$\begin{aligned} A &= \underline{\rho}_i \cdot \dot{\underline{r}} \\ B &= \dot{\underline{R}}_i \cdot \dot{\underline{r}}_s \\ C &= \dot{X}_i^2 + \dot{Y}_i^2 \\ D &= \underline{\rho}_i \cdot (\dot{\underline{R}}_i - \dot{\underline{r}}_s) \\ a &= |\underline{\rho}| \\ b &= \underline{\rho} \cdot \dot{\underline{\rho}} \\ c &= \underline{\rho} \cdot \ddot{\underline{\rho}} + \dot{\underline{\rho}} \cdot \dot{\underline{\rho}} \\ d &= \underline{\rho} \cdot \ddot{\underline{\rho}} + 3\dot{\underline{\rho}} \cdot \dot{\underline{\rho}} \end{aligned} \left. \vphantom{\begin{aligned} A &= \underline{\rho}_i \cdot \dot{\underline{r}} \\ B &= \dot{\underline{R}}_i \cdot \dot{\underline{r}}_s \\ C &= \dot{X}_i^2 + \dot{Y}_i^2 \\ D &= \underline{\rho}_i \cdot (\dot{\underline{R}}_i - \dot{\underline{r}}_s) \\ a &= |\underline{\rho}| \\ b &= \underline{\rho} \cdot \dot{\underline{\rho}} \\ c &= \underline{\rho} \cdot \ddot{\underline{\rho}} + \dot{\underline{\rho}} \cdot \dot{\underline{\rho}} \\ d &= \underline{\rho} \cdot \ddot{\underline{\rho}} + 3\dot{\underline{\rho}} \cdot \dot{\underline{\rho}} \end{aligned}} \right\} \begin{array}{l} \text{(four output quantities)} \end{array}$$

$$\begin{aligned} \rho &= a \\ \dot{\rho} &= \frac{b}{a} \\ \ddot{\rho} &= \frac{c - \dot{\rho}^2}{a} \\ \ddot{\underline{\rho}} &= \frac{d - 3\dot{\rho}\ddot{\rho}}{a} \end{aligned} \left. \vphantom{\begin{aligned} \rho &= a \\ \dot{\rho} &= \frac{b}{a} \\ \ddot{\rho} &= \frac{c - \dot{\rho}^2}{a} \\ \ddot{\underline{\rho}} &= \frac{d - 3\dot{\rho}\ddot{\rho}}{a} \end{aligned}} \right\} \begin{array}{l} \text{(slant range)} \end{array}$$

$$\Delta_{r\rho} = \frac{0.0018958}{(\sin \gamma + 0.06483)^{1.4}} \frac{n_i}{340.0}$$

$$\Delta_{r\rho} = \frac{0.0018958}{\tau} \left[\frac{1}{(\sin F + 0.06483)^{1.4}} - \frac{1}{(\sin G + 0.06483)^{1.4}} \right] \frac{n}{340.0}$$

77-3 of 4
(slant range)

where

- λ_i = longitude station i
- α_G = Greenwich hour angle
- ω = Earth rotation rate
- Δt = light time correction
- ϕ_i = geocentric latitude, station i
- R_i = Earth radius at station i
- \underline{r} = geocentric probe coordinates
- \underline{r}_s = geocentric sun coordinates
- γ = elevation angle
- n_i = index of refraction, station i
- τ = doppler averaging time
- $F = \gamma + \frac{\tau \dot{\gamma}}{2}$
- $G = \gamma - \frac{\tau \dot{\gamma}}{2}$

USE

CALL	RATES	
PZE	RHO	ρ , km
PZE	DRHO	$\dot{\rho}$, km/sec
PZE	DDRHO	$\ddot{\rho}$, km/sec ²
PZE	DDDRHO	$\dddot{\rho}$, km/sec ³
PZE	A	A
PZE	B	B
PZE	C	C
PZE	D	D
PZE	DT	Δt , sec
PZE	I	station index (IX for up-leg, IR for down-leg)

COMMON input:

77-4 of 4

THETAI	θ_i , deg	
GHA	$\dot{\alpha}_G$, deg	
OMEGA	ω , deg/sec	
PHI	ϕ_i , deg	
RI	R_i , km	
XN	\underline{r} , km	} light time corrected
DXN	$\underline{\dot{r}}$, km/sec	
DDXN	$\underline{\ddot{r}}$, km/sec ²	
DDDXN	$\underline{\dddot{r}}$, km/sec ³	
XSUN	$\underline{r}_s, \underline{\dot{r}}_s$	
EL	γ , deg	
FNI	n_i	
TAU	τ , sec	

COMMON output:

RVEC	ρ , km
RDOT	$\dot{\rho}$, km/sec
DRVEC	$\Delta_x \rho$, km
DRDOT	$\Delta_x \dot{\rho}$, km/sec

CODING INFORMATION

Length of subroutine is 435 (10) or 663 (8) words.

REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Anderson, John D., TM 312-409, March 24, 1964.

IDENTIFICATION

78

RCAL

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

RCAL is to be used to compute the coefficients as required in BMATRIX:

$$A = \left(\frac{1}{|\overline{RV} - \overline{R}|^3} - \frac{1}{RV^3} \right)$$

$$B = \left(\frac{3(\overline{RV} \cdot \overline{R} - R^2)}{|\overline{RV} - \overline{R}|^5} \right)$$

$$C = \left(\frac{3}{|\overline{RV} - \overline{R}|^3} \right)$$

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutine used:
SQRT

USE

CALL RCAL(RV) RV, location of the body vector. Results are stored in COMMON locations A, B, and C.

CODING INFORMATION

Length of subroutine is 159 (10) or 237 (8) words.

IDENTIFICATION

79

RCOM

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

RCOM computes the coefficients as defined in BMATRIX:

$$A = \left(\frac{1}{\left| \vec{RV} - \vec{R} \right|^3} \right)$$

$$B = \left(\frac{1}{RV^3} \right)$$

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
SQRT

USE

CALL RCOM(RV) RV, location of the body vector. Results are stored in
COMMON locations A and B.

CODING INFORMATION

Length of subroutine is 91 (10) or 133 (8) words.

IDENTIFICATION

80

READS/ODOFF/MOOPH1
 Melba W. Nead, JPL
 IBM 7094 Fap
 January 4, 1965

PURPOSE

READS is the main program of the link of the ODP which reads data. The routine also includes phasing for all three possible targets, Moon, Mars, and Venus. The target is checked and the phasing is moved so as to be available to the trajectory through the WANT capability of the JPTRAJ monitor.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 - ODATA
 - TAPIO
 - OFFSYS
 - ERROR

USE

Upon the completion of each iteration, the logic of the program causes it to cycle back through READS, all under control of the Source Deck and the JPTRAJ monitor.

CALL ODOFF	To end ODP and return to system as called from ODATA.
MOOPH1	Entry to allow JPTRAJ "WANT" feature to save information from this area of READS, namely, the phasing for SPACE.

CODING INFORMATION

Length of subroutine is 704 (10) or 1300 (8) words.

IDENTIFICATION

81

REAP/REAM

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

To read the combined ephemeris tape as it is written by the subroutine RITEM.

RESTRICTIONS

a. ERROR indication received by encountering EOF on tape

b. Subroutines used:

TAPIO

FLAT

ERROR

METHOD

REAP-REAM read the combined ephemeris tape which has been written by RITEM. The subroutine uses two buffers for efficiency of operation.

USE

CALL REAP Compares TOB with that saved on tape

CALL REAM Compares T with that saved on tape

CODING INFORMATION

Length of subroutine is 1375(10) or 2537(8) words.

IDENTIFICATION

82

REJEC

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

REJEC keeps a record of those residuals which were rejected by the 3-sigma test. Each bit in the rejection table represents a residual. The table is ordered sequentially.

RESTRICTION

COMMON break: 46711

USE

CALL REJEC

PZE I

PZE J

PZE K

where

I = 0 to initialize (set pointer at first bit)

= 1 for normal entry

J = 0 to determine rejection status of current point

= 1 to update rejection status of current point

K = 0	point not rejected	} input if J = 1, output if J = 0
= 1	point rejected	

COMMON input and output:

CAPU 500 word rejection table, with recording capability for 18,000 residuals

CODING INFORMATION

Length of subroutine is 46 (10) or 56 (8) words.

IDENTIFICATION

83

RESID

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

Control program for the link which prints and plots residuals and prints statistics.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

THARP

TYPRYT

IOCS

USE

Upon request for residuals and statistics, this link is called by the JPTRAJ monitor.

CODING INFORMATION

Length of subroutine is 704 (10) or 1300 (8) words.

IDENTIFICATION

84

RESP/COLA
Melba W. Nead, JPL
IBM 7094 Fap
January 4, 1965

PURPOSE

This subroutine is designed to provide both listings and plots of information from the receiving stations. The plots are the residuals, observed minus calculated values, for one, two, or three types versus time. Date, station number, time at start of frame, and pass number are included on each frame. In addition to the above, the listings contain count time, sample time, computed values for the data, and weights. Rejected data points are starred. It is possible to list the residual in fixed (nominal) or floating point (FLRES = 1.0).

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

FGDOUT
CXPLLOT
PKINE
PERNOD
STATID
FIXT
PROUT
FLAPR
BIBCD
ERROR
CDC

A maximum of 8 data residuals can be plotted or listed on one iteration.

USE

CALL RESP With each call to RESP a maximum of 3 residuals is plotted or listed.
The routine must be called again for more; the maximum number is 8.

CODING INFORMATION

Length of subroutine is 883 (10) or 1563 (8) words.

REFERENCE

Trask, D.W., RFP 312-37, Addendum 6, April 4, 1962.

IDENTIFICATION

85-1 of 2

RITEM/WREOF
 Melba W. Nead, JPL
 IBM 7094 Fap
 January 4, 1965

PURPOSE

To write the combined ephemeris tape.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 FLAT
 TAPIO

METHOD

RITEM uses TAPIO to write the combined ephemeris tape. The subroutine uses two buffers alternately to provide efficient timing. Each buffer contains four logical records of 156 words each.

USE

CALL RITEM Writes logical record on tape A4. Each record contains the following information:

T (2)		Time of data point
TOB (2)		Observation time
XJUP (6)		Jupiter vector
XERM (6)		
XMAR (6)		Mars vector
XVEN (6)		Venus vector
XSUN (6)		Sun vector
XMOO (6)		Moon vector
X	}	Probe vector
Y		
Z		
XDOT		
YDOT		
ZDOT		
U (36)		Variational equations
DLO		Nutation in longitude
DOB		Nutation in obliquity

XAC	}		85-2 of 2
YAC			
ZAC			Probe acceleration
XJERK	}		
YJERK			Probe jerk (3rd derivative)
ZJERK			
GMAT (66)		Gamma matrix	

CALL WREOF To write end of file and rewind tape A4 when ephemeris is complete.

CODING INFORMATION

Length of subroutine is 1350(10) or 2506(8) words.

IDENTIFICATION

86

ROT

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

ROT rotates the planetary ephemerides from the mean equator and equinox of 1950.0 to the true equator of equinox of date.

RESTRICTION

COMMON break: 47675

METHOD

$$\begin{pmatrix} x \\ y \\ z \end{pmatrix}_{\text{of date}} = NA \begin{pmatrix} x \\ y \\ z \end{pmatrix}_{1950.0}$$

where

NA = nutation-precession matrix obtained by subroutine PNUT

USE

CALL ROT

COMMON input:

ROTMX NA

XSUN geocentric position and velocity of Sun, 1950.0

XVEN geocentric position and velocity of Venus

XMAR geocentric position and velocity of Mars

XJUP geocentric position and velocity of Jupiter

XMOO geocentric position and velocity of Moon

COMMON output:

XSUN, XVEN, XMAR, XJUP, XMOO rotated to true-of-date

CODING INFORMATION

Length of subroutine is 182 (10) or 266 (8) words.

IDENTIFICATION

87

RSIG

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

The partial of AU is expressed as a sum of terms evaluated for SUN, VENUS, MARS, and JUPITER. RSIG evaluates and sums these terms as they are needed in BMATRIX.

$$\sum_{k=1}^4 GM_k \left[x_k \left(\frac{1}{|r_k - r|^3} - \frac{1}{r_k^3} \right) - \frac{3(r^2 - r \cdot r_k)(x - x_k)}{|r_k - r|^5} \right]$$

where

k = 1 = Sun

k = 2 = Venus

k = 3 = Mars

k = 4 = Jupiter

RESTRICTION

COMMON break: 47675

USE

CALL RSIG(GRA, XPL, SIGX) where GRA is gravity of the body, XPL position of body, and SIGX is the sum. The routine is called for the four bodies and SIGX is accumulated.

CODING INFORMATION

Length of subroutine is 51 (10) or 63 (8) words.

IDENTIFICATION

88

SAVCOM

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

Control for the program to save COMMON. This routine also prints the input matrices when the ODP is initiated and checks the times supplied so the trajectory can be run only for the required period.

RESTRICTIONS

- a. ERROR condition; disk error indicated by DCP
- b. COMMON break: 47675
- c. Subroutines used:

PRIM

TYPRYT

QUEST

MAXIM

ERROR

FLAK

DISCBU

OFFSYS

USE

To save COMMON during the trajectory link, this routine is under control of the JPTRAJ monitor.

CODING INFORMATION

Length of subroutine is 1118 (10) or 2136 (8) words.

IDENTIFICATION

89

SCALE

Melba W. Nead, JPL
Fortran II, Version 3
January 4, 1965

PURPOSE

When the target is Mars or Venus, the target vector is divided by AU as required in calculation of the target centered covariance matrix.

USE

CALL SCALE (XPOS, P) XPOS is the target vector, the quotient is stored in P.

CODING INFORMATION

Length of subroutine is 34 (10) or 42 (8) words.

IDENTIFICATION

90

SORT

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

Time sorted predictions are prepared in Link 4. This is the main program in which these predicts are station sorted and prepared for transmission to the tracking stations.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

THARP

FLAK

TYPRYT

IOCS

USE

Link 5 is called under control of the JPTRAJ Source Deck.

CODING INFORMATION

Length of subroutine is 1069 (10) or 2055 (8) words.

IDENTIFICATION

91-1 of 2

SPHX

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

SPHX converts Earth-fixed spherical coordinates to geocentric equatorial cartesian coordinates. No provision is made for nutations; thus, the output is in the mean equator and equinox of date.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

SIN

COS

GHADP

METHOD

$$x_0 = r_0 \cos \phi_0 \cos(a_G + \lambda_0)$$

$$y_0 = r_0 \cos \phi_0 \sin(a_G + \lambda_0)$$

$$z_0 = W$$

$$\dot{x}_0 = (\dot{U} - \omega V) \cos a_G - (\dot{V} + \omega U) \sin a_G$$

$$\dot{y}_0 = (\dot{U} - \omega V) \sin a_G + (\dot{V} + \omega U) \cos a_G$$

$$\dot{z}_0 = \dot{W}$$

$$U = r_0 \cos \phi_0 \cos \lambda_0$$

$$V = r_0 \cos \phi_0 \sin \lambda_0$$

$$W = r_0 \sin \phi_0$$

$$\dot{U} = v_0(\sin \gamma_0 \cos \phi_0 \cos \lambda_0 - \cos \gamma_0 \sin \gamma_0 \sin \sigma_0 - \cos \gamma_0 \sin \phi_0 \cos \lambda_0 \cos \sigma_0) \quad 91-2 \text{ of } 2$$

$$\dot{V} = v_0(\sin \gamma_0 \cos \phi_0 \sin \lambda_0 + \cos \gamma_0 \cos \gamma_0 \sin \sigma_0 - \cos \gamma_0 \sin \phi_0 \sin \lambda_0 \cos \sigma_0)$$

$$\dot{W} = v_0(\sin \gamma_0 \sin \phi_0 + \cos \gamma_0 \cos \phi_0 \cos \sigma_0)$$

where

$x_0, y_0, z_0, \dot{x}_0, \dot{y}_0, \dot{z}_0$ = geocentric equatorial position and velocity

$r_0, \phi_0, \lambda_0, v_0, \gamma_0, \sigma_0$ = Earth-fixed spherical coordinates

a_G = Greenwich hour angle

ω = Earth rotation rate

USE

CALL SPHX

COMMON input:

RNJ r_0 , km

PHINJ ϕ_0 , deg

THTNJ λ_0 , deg

VNJ v_0 , km/sec

ELNJ γ_0 , deg

AZNJ σ_0 , deg

EPOCH epoch of state vector, d. p. sec past 1950.0

OMEGA ω , deg/sec

COMMON output:

XNJ $x_0, y_0, z_0, \dot{x}_0, \dot{y}_0, \dot{z}_0$ (1 x 6)

CODING INFORMATION

Length of subroutine is 211 (10) or 323 (8) words.

IDENTIFICATION

92

STARP

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

STARP determines whether an observed time point should be deleted or retained on the basis of input data start and stop times.

RESTRICTION

COMMON break: 47675

USE

CALL STARP

PZE I I = 0 for point outside allowed range
= 1 for point within allowed range

COMMON input:

TOB observation time, d. p. sec past 1950.0
PTFD-30 start-stop times

CODING INFORMATION

Length of subroutine is 24 (10) or 30 (8) words.

IDENTIFICATION

93-1 of 3

STPREG

Charles L. Lawson/Terry Kinney, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

Double precision solution of the normal equations of a linear least squares regression problem using the stepwise procedure of M. A. Efroymsen.

This subroutine is particularly intended for use in least-squares problems in which the user is aware of a large number of basic functions which might bear a linear relationship to the object function but expects that only some of these basic functions make a significant contribution to the determination of the object function.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
(DFAD)/(DFMP)/(DFDP)
DSQRT

METHOD

In an n-parameter problem this subroutine provides successively a 1-parameter fit, a 2-parameter fit, . . . , a k-parameter fit, where the new parameter introduced at each step is the one which will make the greatest reduction in the sum of squares of residuals. This criterion is equivalent to the requirement that the new parameter be selected as the one which will have the highest statistical significance in the sense that the ratio of its magnitude to its standard deviation will be largest or equivalently its "F" value will be largest (see Ref. c). The termination level k, which will be $\leq n$, is determined by the subroutine on the basis of tolerance parameters SDIN, SDOUT, and CD provided by the user.

For an exposition of this procedure, see Ref. a of this subroutine. For a complete single precision regression package which incorporates this procedure in its system solving subroutine, see Ref. b. For a discussion of this procedure including details of its implementation in this program see Ref. c.

USE

Let the overdetermined system of linear equations which expresses the linear regression problem be denoted by

$$Gc = f$$

where G is an m by n matrix ($m > n$), c is an n-vector, f is an m-vector. The elements of G and f are known and the coefficient vector c is to be computed as a weighted least squares solution.

The normal equations $Ac = b$ are formed from the overdetermined system by pre-multiplication by G^*W , (*denotes transposition) where W is an m by m non-negative definite weighting matrix (W may be the identity matrix).

This computation, as well as the computation of the scalar quantity $s = f^*Wf$, is to be done by the user's program. STPREG may then be called as follows:

```
D  DIMENSION A (20, 20), B (20), S (1)
   DIMENSION INOUT (20)
   CALL STPREG (A, B, S, N, M, SDIN, SDOUT,
               CD, NIN, INOUT)
```

The parameters are defined as follows:

A(I, J), I = 1, N; J = 1, N

On entry: The matrix A, computed by the user as $A = G^*WG$.

A is normalized by STPREG before inversion.

On return: A^{-1} if $NIN = N$. If $NIN < N$ then only an NIN by NIN submatrix of A will have been inverted and the elements of this inverse will be in the locations A (I, J) corresponding to those values of I and J for which both $INOUT(I) = +1$ and $INOUT(J) = +1$. The other A (I, J) locations for $I \leq N, J \leq N$ will contain zero. A^{-1} is unnormalized before the return.

B(I), I = 1, N

On entry: The vector b, computed by the user as $b = G^*Wf$.

On return: The solution vector c if $NIN = N$. If $NIN < N$ then only NIN components of c will have been computed and they will be those in the locations B (I) for which $INOUT(I) = +1$. The other B (I) locations for $I \leq N$ will contain zero.

S

On entry: The scalar s, computed by the user as $s = f^*Wf$.

On return: The weighted sum of squares of residuals:

$$\begin{aligned} (f - Gc)^*W(f - Gc) &= (f - Gc)^*Wf \\ &= f^*Wf - (Gc)^*Wf = f^*Wf - c^*b \end{aligned}$$

N The dimension, n, of the normal system.

M The dimension, m, of the f vector.

SDIN, SDOUT

If $DSIN > 0.$, these two numbers will be used by the "significance" logic of the program. A coefficient will be permitted to enter the solution if the ratio of its

magnitude to its standard deviation will exceed SDIN. If this ratio subsequently drops below SDOUT this coefficient will be removed from the solution. To avoid looping, SDIN should exceed SDOUT.

Typical values for SDIN and SDOUT might be SDIN = 1.96, SDOUT = 1.64, which could be interpreted as meaning that a coefficient will be permitted to enter the solution if there is a 95% probability that it is different from zero and will later be removed from the solution if this probability falls below 90%.

If $SDIN \leq 0$ all significance tests will be skipped. This permits use of the program with meaningless data in B and S, as may be the case when only matrix inversion is desired.

CD Tolerance for relative reduction in magnitude of diagonal elements. The i^{th} coefficient will not be entered into the solution if the i^{th} diagonal element becomes smaller than CD times the original i^{th} diagonal element of the matrix A. For example if CD = 10. ****(-12)** then a potential pivot element will not be used if it has lost more than its first 12 significant decimal places.

NIN On return: The number of coefficients in the final solution. INOUT (I), I = 1, N
On return: Flags to indicate final status of each coefficient.

- +1 Included in solution.
- 1 Rejected by diagonal test.
- 2 Rejected by significance tests.

CODING INFORMATION

Length of subroutine is 1069 (10) or 2055 (8) words.

REFERENCES

- a. Efroymson, M. A., Multiple Regression Analysis, Mathematical Methods for Digital Computers, ed. Ralston and Wilf, Wiley, 1960, pp. 191-203.
- b. Efroymson, M. A., Stepwise Multiple Regression with Variable Transformations, SHARE Distribution No. 1194, October, 1961.
- c. Lawson, C.L., Computation of the Most Significant Coefficients in Least-Squares Estimation, JPL Section Report No. 372-5, August, 1962.

IDENTIFICATION

94-1 of 2

TACCOM

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

To evaluate the target-centered cartesian Covariance Matrix.

RESTRICTIONS

a. COMMON break: 47055

b. Subroutines used:

AAT

SCALE

(DFAD)

(DFMP)

(DFSB)

METHOD

In the following, A will represent the ephemeris scaling factor. Thus, A = REM when target = MOON, and A = AU when target = MARS or VENUS.

a. When A is among the estimated parameters, compute the target centered covariance matrix as

$$\Gamma_E = \Gamma_X + PP^T \sigma_A^2 - \Gamma_{XA} P^T - P \Gamma_{XA}^T$$

where

Γ_X = upper left hand 6×6 sub-matrix of the geocentric covariance matrix Γ

P = 6-vector giving the geocentric position and velocity of the target in units of A and A/sec, referred to the Earth equatorial system

σ_A^2 = diagonal element of Γ corresponding to A

Γ_{XA} = first 6 elements of the column of Γ corresponding to A

b. When A is not among the estimated parameters

94-2 of 2

$$\Gamma_E = \Gamma_X$$

USE

CALL TACCOM (GAME, ITAR)
GAME contains Γ_E as calculated in TACCOM
ITAR contains the target reference
 ITAR = 1 MARS
 ITAR = 2 MOON
 ITAR = 3 VENUS

CODING INFORMATION

Length of subroutine is 938 (10) or 1652 (8) words.

REFERENCE

Carey, C., RFP 312-252, March 5, 1964.

IDENTIFICATION

95

TAMP/EPOX

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

Writes mapping times on scratch tape A7 to be saved for matrix manipulation program input tape.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

TAPIO

FLAT

USE

CALL TAMP

EPOX

provides space for storage of injection conditions which are saved through the WANT cards feature of JPTRAJ

CODING INFORMATION

Length of subroutine is 87 (10) or 127 (8) words.

IDENTIFICATION

96

THARP/PKINE

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

THARP serves as the logical control routine for the predictions and residuals output links, LA5 and LA8. It sets up lists of desired stations and calls the output subroutines AQU1, PERNOD, RESP, and ANGPLT. PKINE is called by these output routines in order to obtain a meaningful logical record from the disk read routine KINE.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 - OFFSYS
 - KINE
 - AQUI/ALLI/BAQUI/BALLI
 - PERNOD/WOLF
 - RESP
 - ANGPLT
 - BIBCD
 - TYPRYT

USE

CALL THARP
 CALL PKINE
 PZE SUBR = 1 if called from ANGPLT
 = 2 if called from AQU1
 = 3 if called from PERNOD
 = 4 if called from RESP

COMMON input:

IF3 ODP option flags
 IRES residual output flags (8 × 15)
 RESR plotting scale factors (8 × 15)
 ITNO iteration number
 PT pointing times (used as flags)

CODING INFORMATION

Length of subroutine is 379 (10) or 573 (8) words.

IDENTIFICATION

97-1 of 2

TIMER

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

TIMER obtains the primary light time correction which adjusts the observation time for ephemeris lookup purposes.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 - (DFSB)
 - LOOKUP
 - SQRT

METHOD

$$\Delta t = 0$$

$$t = t_{ob} - \Delta t$$

$$r = \sqrt{x^2 + y^2 + z^2}$$

$$\Delta t = \frac{r - R_e}{C}$$

The procedure is then repeated with the new Δt .

USE

CALL TIMER

COMMON input:

TOB t_{ob} , d. p. sec past 1950.0

RE R_e , km

VELC c , km/sec

COMMON output:

97-2 of 2

T t, d. p. sec past 1950.0

DELTA Δt , d. p. sec

CODING INFORMATION

Length of subroutine is 98 (10) or 142 (8) words.

IDENTIFICATION

98

TNORM

Melba W. Nead/Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

TNORM normalizes an $N \times N$ matrix on its diagonal terms. N may be 2, 3, 4, 5, or 6 but the matrix must be stored in a 6×6 array.

RESTRICTION

Subroutine called:

 SORT

USE

CALL TNORM

PZE YMAT input matrix

PZE XMAT output normalized matrix

PZE N N

CODING INFORMATION

Length of subroutine is 170 (10) or 252 (8) words.

IDENTIFICATION

99-1 of 3

TOCIM

Michael R. Warner, JPL

IBM 7094, Fap

January 4, 1965

PURPOSE

TOCIM computes occultation time and impact time, and obtains the residual. It writes a record on the combined ephemeris tape for each observable, and outputs on disk for subsequent printing.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

OFFSYS

TIMER

LOOKUP/KOOL

INTRI

SQRT

COS/SIN

GAMAT

GHADP

ARCOS/ARSIN

FLAT

RITE

WRITEB

METHOD

TOCIM uses a Newton-Raphson formula for both data types:

For occultation time,

$$\left[T_0^{(r)} - T_0^{(r-1)} \right] \frac{d(\phi_p - \phi_m)}{dt} \Big|_{T_0^{(r-1)}} = \phi_m - \phi_p$$

where

$$\phi_m = \sin^{-1} \frac{R_m}{\rho_m}$$

$$\phi_p = \cos^{-1} \frac{\underline{\rho} \cdot \underline{\rho}_m}{|\underline{\rho}| |\underline{\rho}_m|}$$

$$\frac{d(\phi_p - \phi_m)}{dt} = \frac{\underline{\rho} \cdot \underline{\rho}_m}{a_1} \left[\left(\frac{a_2}{\rho^2} + \frac{a_3}{\rho_m^2} \right) - a_4 \right] + \frac{a_3 R_m}{\rho_m^2 \sqrt{\rho_m^2 - R_m^2}}$$

$$a_1 = |\underline{\rho}| |\underline{\rho}_m| \sin \phi_p$$

$$a_2 = \underline{\rho} \cdot \dot{\underline{\rho}}_m$$

$$a_3 = \underline{\rho}_m \cdot \dot{\underline{\rho}}_m$$

$$a_4 = \underline{\rho} \cdot \dot{\underline{\rho}}_m + \underline{\rho}_m \cdot \dot{\underline{\rho}}$$

For impact time,

$$\left[T_I^{(r)} - T_I^{(r-1)} \right] \left| \dot{\underline{r}} - \dot{\underline{r}}_m \right| = \left| \underline{r} - \underline{r}_m \right| - R_m$$

(all quantities are defined in the subroutine OCPAR documentation)

USE

CALL TOCIM

COMMON input:

DXDK-6	target identification
IFA-4	R_m , km
PTFD	occultation-impact data buffer
DXDR-27	occultation-impact buffer
LLIST	physical constants estimate flags
RI	station Earth radius, km
PHI	station geocentric latitude, deg
THETA	station longitude, deg
EPOCH	ODP epoch, d. p. sec past 1950.0
OMEGA	Earth rotation rate, deg/sec

CODING INFORMATION

99-3 of 3

Length of subroutine is 998 (10) or 1746 (8) words.

REFERENCE

Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.

IDENTIFICATION

100-1 of 9

UMAT

Kenneth Oslund/Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

PURPOSE

UMAT computes the encounter conic, rotating from the geocentric equatorial system to the target-centered \underline{R} , \underline{T} system. All encounter parameters and statistics are obtained by this routine.

RESTRICTIONS

Subroutines called:

SQRT

LOG

GERPU

ARTAN

PARAM

METHOD

The following inputs are in the UMAT calling sequence:

\underline{r}'_{tp} , $\underline{\dot{r}}'_{tp}$ target centered probe position and velocity at time of encounter

\underline{r}_{ct} , $\underline{\dot{r}}_{ct}$ position and velocity of target, centered at that body's primary, at time of encounter

Γ_e covariance matrix at encounter (7×7)

U_e mapping matrix to encounter (7×7)

μ GM_{target}

t_f time of flight

First, the rotation matrix is obtained:

$$r'^2 = \underline{r}_{ct} \cdot \underline{r}_{ct}$$

$$\dot{s}'^2 = \underline{\dot{r}}_{ct} \cdot \underline{\dot{r}}_{ct}$$

$$\mathbf{r}' \dot{\mathbf{r}}' = \mathbf{r}_{ct} \cdot \dot{\mathbf{r}}_{ct}$$

$$\underline{u}_p = \frac{\mathbf{r}_{ct}}{r}$$

$$\underline{v}_p = \frac{r' \dot{\mathbf{r}}_{ct} - \dot{r}' \mathbf{r}_{ct}}{r' \sqrt{\dot{s}'^2 - \dot{r}'^2}}$$

$$a_{33} = \sqrt{\left(u_{pr} + v_{py}\right)^2 + \left(u_{py} - v_{px}\right)^2} - 1$$

$$a_{23} = \sqrt{u_{pz}^2 + v_{pz}^2}$$

$$a_{11} = \frac{\frac{1}{1 + a_{33}} \left[v_{pz}(u_{px} + v_{py}) + u_{pz}(u_{py} - v_{px}) \right]}{a_{23}}$$

$$a_{12} = \frac{\frac{1}{1 + a_{33}} \left[v_{pz}(u_{py} - v_{px}) - u_{pz}(u_{px} + v_{py}) \right]}{a_{23}}$$

$$a_{21} = -a_{12} a_{33}$$

$$a_{22} = a_{11} a_{33}$$

$$a_{31} = a_{12} a_{23}$$

$$a_{32} = -a_{11} a_{23}$$

The target-centered probe vector is then rotated:

$$\underline{\mathbf{r}}_{tp} = A \underline{\mathbf{r}}'_{tp}$$

$$\underline{\dot{\mathbf{r}}}_{tp} = A \underline{\dot{\mathbf{r}}}'_{tp}$$

and the following column vectors are formed:

100-3 of 9

$$\underline{X}_p = \begin{pmatrix} r_{tp} \\ 0 \end{pmatrix} \quad \dot{\underline{X}}_p = \begin{pmatrix} \dot{r}_{tp} \\ 0 \end{pmatrix}$$

$$\underline{X}_v = \begin{pmatrix} 0 \\ r_{tp} \end{pmatrix} \quad \dot{\underline{X}}_v = \begin{pmatrix} 0 \\ \dot{r}_{tp} \end{pmatrix}$$

(the seventh member of these vectors is in each case zero)

The encounter conic is then obtained:

$$r^2 = \underline{X}_p \cdot \underline{X}_p$$

$$\dot{s}^2 = \dot{\underline{X}}_p \cdot \dot{\underline{X}}_p$$

$$r \dot{r} = \underline{X}_p \cdot \dot{\underline{X}}_p$$

$$a = \frac{1}{\frac{2}{r} - \frac{\dot{s}^2}{\mu}} \quad (\text{semi-major axis})$$

$$e^2 = 1 - \frac{r^2 \dot{s}^2 - (r \dot{r})^2}{\mu/a} \quad (\text{eccentricity})^2$$

$$e \sinh F = \frac{r \dot{r}}{\sqrt{-\mu/a}}$$

$$e \cosh F = 1 - \frac{r}{a}$$

$$e \exp F = e \sinh F + e \cos F$$

$$F = \log (\exp F) \quad (\text{hyperbolic anomaly})$$

$$e p_z = X_{pz} \frac{e \cosh F}{r} - \dot{X}_{pz} e \sinh F \sqrt{-\frac{a}{\mu}}$$

$$e \sin \phi_s = \left(\frac{X_{pz}}{r} - X_{pz} \sqrt{-\frac{a}{\mu}} \right) \exp F + \dot{X}_p e \sqrt{-\frac{a}{\mu}}$$

$$\underline{B} \cdot \underline{T} \cos \phi_s = \sqrt{-\frac{a}{\mu}} (X_{pz} \dot{X}_{py} - X_{py} \dot{X}_{pz})$$

$$\underline{B} \cdot \underline{R} \cos \phi_s = a(\sin \phi_s - e p_z)$$

$$T_a = \frac{3}{2} \sqrt{-\frac{a}{\mu}} (e \sinh F - F + \log e)$$

$$T_e = \left(\frac{e \sinh F + 1}{e} \right) a \sqrt{-\frac{a}{\mu}}$$

$$T_F = -r \sqrt{-\frac{a}{\mu}}$$

$$v_\infty = \frac{1}{\sqrt{-\frac{a}{\mu}}}$$

$$n = \frac{-v_\infty}{a}$$

$$t_L = t_f + \frac{(F - e \sinh F - \log e)}{n} \quad (\text{linearized time of flight})$$

The next step is to obtain the encounter partials. The encounter parameter set (M) is

$\underline{B} \cdot \underline{R}$ as defined above

$\underline{B} \cdot \underline{T}$ as defined above

t_L linearized time of flight

$\left. \begin{array}{l} \underline{S} \cdot \underline{R}_s \\ \underline{S} \cdot \underline{T}_s \end{array} \right\}$ dot products of incoming asymptote unit vector \underline{S}
 with standard \underline{R} and \underline{T} vectors

C_3 vis-viva energy

These parameters are related to the position and velocity of the probe at encounter and the target gravitational constant. This set is called Q.

$$F_{az} = \frac{\exp F}{r}$$

$$F_{a\dot{z}} = -\sqrt{-\frac{a}{\mu}} (\exp F - e)$$

$$F_{bz} = \frac{e \cosh F}{r}$$

$$F_{b\dot{z}} = -\sqrt{-\frac{a}{\mu}} e \sinh F$$

$$F_{cx} = -\dot{X}_{py}$$

$$F_{cy} = \dot{X}_{px}$$

$$F_{c\dot{x}} = X_{py}$$

$$F_{c\dot{y}} = -X_{px}$$

$$S_x = \left[\left(\frac{X_{px}}{r} - \sqrt{-\frac{a}{\mu}} \dot{X}_{px} \right) \exp F + e \sqrt{-\frac{a}{\mu}} \dot{X}_{px} \right] \frac{1}{e} x \rightarrow y$$

$$S_z = \sin \phi_s$$

$$T_{sx} = \frac{S_y}{\cos \phi_s}$$

$$T_{sy} = -\frac{S_x}{\cos \phi_s}$$

$$R_{sx} = T_{sy} S_z$$

$$R_{sy} = -T_{sx} S_z$$

$$R_{sz} = \cos \phi_s$$

$$\underline{\delta a} = 2a^2 \left(\frac{X_p}{r^3} + \frac{\dot{X}_n}{\mu} \right)$$

$$\delta a_\mu = - \frac{a^2(1 + e \cosh F)}{r\mu}$$

$$\left[\underline{\delta e}; \delta' e_\mu \right] = \frac{1}{e} \left[- \frac{\frac{e}{a} \cosh F \underline{X}_p}{r} - \frac{e \sinh F (\dot{\underline{X}}_p + \underline{X}_v)}{\sqrt{-\frac{\mu}{a}}} + \delta \underline{a} \left(\frac{\frac{1}{2} e^2 \sinh^2 F}{a} + \frac{r e \cosh F}{a^2} \right) \right]$$

$$\delta e_\mu = \delta' e_\mu + \frac{1}{e} \left[- \frac{1}{\mu} \left(\frac{1}{2} e^2 \sinh^2 F + \frac{r}{a} e \cosh F \right) + \frac{r^2}{2a^2} + \frac{r^2 \delta a_\mu}{2a^3} \right]$$

$$\left[\underline{\delta F}; \delta' F_\mu \right] = \frac{1}{e} \left[\frac{\frac{e}{a} \sinh F \underline{X}_p}{r} + \frac{e \cosh F (\dot{\underline{X}}_p + \underline{X}_v)}{\sqrt{-\frac{\mu}{a}}} - \delta \underline{a} \left(\frac{\frac{1}{2} e^2 \cosh^2 F}{a} + \frac{r e \sinh F}{a^2} \right) \right]$$

$$\delta F_\mu = \delta' F_\mu + \frac{1}{e} \left(- \frac{e \sinh F}{2\mu e \cosh F} - \frac{r^2 e \sinh F \delta a_\mu}{2 e \cosh F a^3} \right)$$

$$\left[\underline{\delta \Phi_s}; \delta' \Phi_{s\mu} \right] = \left[- \frac{\underline{X}_{pz} \underline{X}_p \exp F}{r^3} - \frac{\frac{1}{2} \dot{\underline{X}}_{pz} (e - \exp F) \partial \underline{a}}{\sqrt{-\frac{\mu}{a}}} \left(- \frac{\dot{\underline{X}}_{pz} a}{\sqrt{-\frac{\mu}{a}}} + \sin \Phi_s \right) \delta \underline{e} - \left(\dot{\underline{X}}_{pz} \sqrt{-\frac{a}{\mu}} - \frac{\underline{X}_{pz}}{r} \right) \delta \underline{F} \exp F + \frac{F}{a} \right] \frac{1}{e \cos \Phi_s}$$

$$\left[\underline{\delta S_x}; \delta' S_{x\mu} \right] = \frac{1}{e} \left[- \frac{\underline{X}_{px} \exp F \underline{X}_p}{r^3} + \frac{\dot{\underline{X}}_{px} (\exp F - e)}{2\sqrt{-\frac{\mu}{a}}} \delta \underline{a} + \left(\sqrt{-\frac{a}{\mu}} \dot{\underline{X}}_{px} - S_x \right) \delta \underline{e} - \left(\frac{\sqrt{-\frac{a}{\mu}} \dot{\underline{X}}_{px} - \underline{X}_{px}}{r} \right) \exp F \delta \underline{F} + \frac{F}{a} \right]$$

$$\left[\underline{\delta S}_y \vdots \delta' S_{y\mu} \right] = \left[\underline{\delta S}_x \vdots \delta' S_{x\mu} \right] \quad x \rightarrow y$$

$$\left[\underline{\delta S}_z \vdots \delta' S_{z\mu} \right] = \cos \phi_s \underline{\delta \phi_s}$$

$$\underline{E}_p = - \frac{e \sinh F \sqrt{-\frac{a}{\mu}} \dot{X}_p}{2\mu}$$

$$\underline{\delta S}_\mu = \underline{\delta' S}_\mu + \left[\underline{E}_p - (e \cosh F - e^2) \frac{\dot{X}_p \sqrt{-\frac{a}{\mu}}}{2\mu} \right] \frac{1}{e \cos \phi_s}$$

$$\delta \phi_{s\mu} = \delta' \phi_{s\mu} + \delta' S_{z\mu}$$

$$\left[e \delta P_z \vdots (e \delta P_{z\mu}) \right] = - \frac{X_{pz} e \cosh F X_p}{r^3} + \frac{X_{pz} e \sinh F \delta a}{2\sqrt{-\frac{\mu}{a}}}$$

$$\left(- \dot{X}_{pz} e \cosh F \sqrt{-\frac{a}{\mu}} - \frac{e \sinh F}{r} \right) \underline{\delta F} + \underline{F}_b$$

$$e \delta P_{z\mu} = (e \delta P_{z\mu}) + E_{pz}$$

The program now computes the variational matrix $V = \left(\frac{\partial M}{\partial Q} \right)$:

$$\left[\underline{v}_1 \vdots v'_{1\mu} \right] = \left[\left(e \sin \phi_p - e^2 P_z \right) \underline{\delta a} - a e P_z \underline{\delta e} - (\underline{B} \cdot \underline{R} e \sin \phi_p a e \cos \phi_p) \underline{\delta \phi_s} - a e^2 \underline{\delta P_z} \right] \frac{1}{e \cos \phi_p}$$

$$\left[\underline{v}_2 \vdots v'_{2\mu} \right] = \left(\frac{e \cos \phi_p \underline{B} \cdot \underline{T}}{2a} \underline{\delta a} + \underline{B} \cdot \underline{T} e \sin \phi_p \underline{\delta \phi_s} - e \sqrt{-\frac{a}{\mu}} \underline{F}_c \right) \frac{1}{e \cos \phi_p}$$

$$\left[\underline{v}_3 \vdots v'_{3\mu} \right] = T_a \underline{\delta a} + T_e \underline{\delta e} + T_F \underline{\delta F}$$

$$\left[\underline{v}_4 \vdots v'_{4\mu} \right] = \frac{\mu}{a^2} \underline{\delta a}$$

$$\left[\underline{v}_5 \vdots v'_{5\mu} \right] = T_{sx} \underline{\delta S}_x + T_{sy} \underline{\delta S}_y$$

$$\begin{bmatrix} v_6 \\ v_6 \end{bmatrix}' = R_{sx} \delta S_x + R_{sy} \delta S_y + R_{sz} \delta S_z$$

100-8 of 9

$$v_{1\mu} = v_{1\mu}'$$

$$v_{2\mu} = v_{2\mu}' + \frac{\underline{B} \cdot \underline{T}}{2\mu}$$

$$v_{3\mu} = v_{3\mu}' - \frac{\sqrt{-\frac{a^3}{\mu}} (F - e \sinh F - \log e)}{2\mu}$$

$$v_{4\mu} = v_{4\mu}' - \frac{1}{a}$$

$$v_{5\mu} = v_{5\mu}'$$

$$v_{6\mu} = v_{6\mu}'$$

The following matrix manipulations are then performed:

$$N = V^T A \Gamma_e A^T V \quad (\text{noise moment matrix})$$

$$V_0 = V A U_e \quad \left(V_0 = \frac{\partial M}{\partial Q_0} \text{ matrix} \right)$$

Finally, the configuration of the miss ellipse is obtained:

$$r = \sqrt{\left(\frac{n_{11} - n_{22}}{2} \right)^2 + n_{12}^2}$$

$$s = \frac{n_{11} + n_{22}}{2}$$

$$a_m = \sqrt{s + r} \quad (\text{semi-major axis})$$

$$b_m = \sqrt{s - r} \quad (\text{semi-minor axis})$$

$$\theta = \frac{1}{2} \tan^{-1} \left(\frac{2n_{12}}{n_{22} - n_{11}} \right) \quad (\text{inclination})$$

USE

100-9 of 9

CALL	UMAT		
PZE	XT	$\dot{r}_{tp}, \dot{i}_{tp}$	(1 × 6)
PZE	XH	$\dot{r}_{ct}, \dot{i}_{ct}$	(1 × 6)
PZE	G	Γ_e	(7 × 7)
PZE	U	U_e	(7 × 7) in (20 × 20) array
PZE	TMU	μ of target	
PZE	TF	t_f , sec	

CODING INFORMATION

Length of subroutine is 1681 (10) or 3221 (8) words.

REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Anderson, John D., RFP 312-83, October 1962.
- c. Null, George W., RFP 312-179, Addendum 2, September 19, 1963.

IDENTIFICATION

101

VEC

Charles Coltharp, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

VEC computes the standard dot product of two 1×3 vectors, $A = \underline{X} \cdot \underline{Y}$.

RESTRICTION

COMMON break: 47675

USE

CALL VEC

PZE X

PZE Y

COMMON output:

A dot product

CODING INFORMATION

Length of subroutine is 15 (10) or 17 (8) words.

IDENTIFICATION

102-1 of 2

WAIT
 Michael R. Warner, JPL
 Fortran II, Version 3
 January 4, 1965

PURPOSE

WAIT obtains the a priori weight for each data point used in the normal equations.

RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
 - COL
 - DECOD
 - SQRT
 - WOCT
 - COS
 - ERROR

Error condition:

If the weight obtained is less than 10^{-20} , the program will abort.

METHOD

Method a1.

If a priori coefficients $\sqrt{\tilde{\omega}_j}$ are input to the program, the calculated weight is

$$\sqrt{\omega_j} = \sqrt{\tilde{\omega}_j} \beta_1 \beta_2 \sqrt{\frac{60}{\tau_s}}$$

where

$$\beta_1 = 1 \text{ when not weighting azimuth or hour angle}$$

$$= \frac{1}{\cos \gamma} \text{ when weighting azimuth angle}$$

$$= \frac{1}{\cos \delta} \text{ when weighting hour angle}$$

$$\beta_2 = 1 \text{ when weighting azimuth angle}$$

$$= 1 + \frac{18}{(\gamma + 1)^2} \text{ when not weighting azimuth angle}$$

τ_s = data sample rate

102-2 of 2

δ = declination

γ = elevation angle

Method a2.

If octal weight codes are input to the ODP a priori coefficient table, the program uses Method b.

Method b.

If octal weight codes are input by the ODG data file, and if no input was made to the ODP coefficient array, the calculated sigma is

$$\omega_j = \sum_{i=1}^6 S_{pjk}^2 g_{pj}^2 \max \left(\frac{T_{pjk}}{\tau}, 1 \right)$$

(see documentation for subroutines DECOD and COL).

USE

CALL WAIT

COMMON input:

BALNC $\sqrt{\omega_j}$ (12 x 15)

ID data type identification

IR receiver identification

SSQ S_{pjk}^2 (12 x 6 x 4)

TL T_{pjk} (12 x 6 x 4)

GSQ g_{pjk}^2 (6)

TAU τ (doppler averaging time, sec)

IF10 phi vector output flag

DEC δ , deg

EL γ , deg

TS τ_s , sec

COMMON output:

RVECW array of weights

CODING INFORMATION

Length of subroutine is 230 (10) or 346 (8) words.

REFERENCE

Hamilton, Thomas W., TM 312-182, April 12, 1962.

IDENTIFICATION

103

WASH

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

WASH prints certain time-dependent quantities which are calculated by the ODP (see list below). The subroutine is operative only in Mode-IV or the JPTRAJ production mode.

RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

READB

FLAT/FLAPR

FIXT

PROUT

USE

CALL WASH

COMMON input:

TOB observation time, d. p. sec past 1950.0

GHA Greenwich hour angle, deg

X geocentric probe vector

XSUN geocentric Sun vector

XMOO geocentric Moon vector

XVEN geocentric Venus vector

XMAR geocentric Mars vector

CPPHI partials $\partial F/\partial Q$

CPTHHT partials $\partial F/\partial \tilde{Q}$

U partials $\partial \underline{r}/\partial \underline{r}_0$

GMAT partials $\partial \underline{r}/\partial q$

DRVEC refraction corrections

CODING INFORMATION

Length of subroutine is 401 (10) or 621 (8) words.

IDENTIFICATION

104

WOCT

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

PURPOSE

WOCT determines whether an input a priori weight represents a floating point quantity or an octal weight code.

USE

CALL	WOCT	
PZE	WT	a priori weight word
PZE	CODE	output if octal code
PZE	TEST	= 0 if floating point ≠ 0 if octal code

CODING INFORMATION

Length of subroutine is 11 (10) or 13 (8) words.

VII. OPERATING INSTRUCTIONS

The ODP may be operated in three modes: SFOF Mode II, SFOF Mode IV, and the JPTRAJ Production Mode. Mode II requires the full SFOF hardware configuration, namely, the 7094-disk-7040, with associated user-area equipment. Mode IV requires the 7094 and disk, properly initialized for SFOF operation. The JPTRAJ Production Mode is independent of the SFOF system, requiring only the 7094 and disk.

A. MODE II OPERATION

1. After the system has been initialized and the user programs loaded on disk, ask the operator to mount the following tapes:

- A3 scratch for 1401 output
- A4 scratch for combined ephemerides
- B6 EPHEM system tape
- B9 scratch for 4020 plot output (optional)

2. Type PS1X\$ at the user area 7 console.

3. Depress the TURN ON button, then the TRANSMIT MESSAGE button.

4. When requested by the administrative printer, enter the ODPX source deck via the user area card reader.

5. At the conclusion of PS1X, type ODPX\$ at the user-area 5 console.

6. Depress the TURN ON button, then the TRANSMIT MESSAGE button.

7. At the beginning of ODPX execution, the message ENTER SWITCHES. 22 = TERMINATE, 23 = DELAY, NONE = PROCEED will be printed by the administrative printer. To proceed, clear the switches and type ODPX\$ at the user area console.

8. Depress the OPTION SWITCH ENTRY button, then the TRANSMIT MESSAGE button.

9. When requested by the administrative printer, enter the input deck, as described in Section IV A, via the user area card reader. The program will then execute the options requested by the input deck.

10. To obtain the full SC-3070 printout, type 1\$ at the console, depress the AUTO ON button, then the TRANSMIT MESSAGE button.

11. The message described in step 7 will be printed between iterations. The user has the opportunity to read a new deck for the next iteration, or terminate ODP execution.

B. MODE IV OPERATION

1. After the system has been initialized and the user programs loaded on disk, mount the following tapes:

- A3 scratch for 1401 output
- A4 scratch for combined ephemeris
- B6 EPHEM system tape
- B7 scratch for MMP output (optional)
- B1 scratch for supplementary partials output (optional)
- B9 scratch for 4020 plot output (optional)

2. While the system is at the idle stop, depress sense switch 1 and depress START at the 7094 console.

3. At the card reader select, enter the card TURN ON, PS1X\$.

4. Raise sense switch 1.

5. At the card reader select, enter the ODPX source deck.

6. At the conclusion of PS1X (idle stop), depress sense switch 1 and depress START.

7. At the card reader select, enter the card TURN ON, ODPX\$.

8. Raise sense switch 1.

9. At the card reader select, enter the input deck, as described. The program will then execute the options requested by the input deck.

10. To obtain on-line printout, depress sense switch 6.

11. The card reader will select between iterations for additional input. If console key 22 is depressed at the end of an iteration, the execution will be terminated.

C. JPTRAJ PRODUCTION MODE OPERATION

1. Mount the following tapes:

- A1 JPTRAJ system tape
- A2 input tape (see below)
- A3 scratch for 1401 output
- A4 scratch for combined ephemeris
- A5 JPTRAJ program tape
- B1 scratch for system operation
- B6 EPHEM system tape
- B7 scratch for MMP output (optional)
- B9 scratch for 4020 plot output (optional)

2. The input tape should contain the following card images:

```

$JOB MRW, 2116000, 542-70285-1-3170, FC (example of
      JPL job card)
*      JPTRAJ
$RESTORE
*      DATA
      [ ODPX source deck ]
      [ ODPX input deck ]
      [   for iteration 1 ]
      .
      .
      .
      .
      .
      [ ODPX input deck ]
      [   for iteration n ]
      EOF card
$PAUSE
    
```

3. To start the system, depress the LOAD TAPE button at the 7094 console.
4. Once the program has been restored on disk, the ~~RESTORE~~ card and the A5 tape need not be used.

D. SOURCE DECK LISTING

JPL TECHNICAL MEMORANDUM NO. 33-204

ANALYSIS OF THE JPTRAJ SOURCE PROGRAM FOR THE ODP PROVIDES THE BASIC LOGIC OF THE PROGRAM. IN THE FOLLOWING THE STARRED ITEMS ARE DIRECTLY FROM THE SOURCE PROGRAM WHICH IS GIVEN IN COMPLETE FORM IN THE APPENDIX. NON-STANDARD EXITS FOR EACH LINK ARE EXPLAINED. THE STANDARD EXIT IS IN SEQUENCE.

```

*V1  ODPX  N9
      N9

*V9  LA9   X,N11,N1,N4
      X    TERMINATE JOB
      N11  USE SAME PROBE EPHEMERIS, SKIP TRAJ THIS PASS
      N1   RESTART FEATURE, TO RESET TO NOMINAL INPUT
      N4   USE SAME COMBINED EPHEMERIS, SKIP TO LINK 4
           THE CARD SWS=1 WILL CAUSE THE ODP TO READ THE INPUT
           DECK FROM TAPE A2. NORMAL INPUT IS BY ONLINE READER.

*V12 LA12  X
      X    TERMINATE JOB

*V13 SPACE X
      X    TERMINATE JOB

*V11 LA11  N6,N6,N12
      N4   EXIT TO PREDICT-DATA TAPE LINK
      N6   EXIT TO DATA FITTING LINK
      N12  EXIT TO LINK 12 FOR ERROR, THEN TO X

*V2  LA2   N12, X
      N12  EXIT TO LINK 12 FOR ERROR COMMENT, THEN TO X
      X    TERMINATE JOB

*V2A LA2A  N9,N12
      N9   STEP MAP COMPLETED, CHECK FOR MORE INPUT
      N12  EXIT TO LINK 12 FOR ERROR, THEN TO X

*V3  LA3   N9,N10
           (NO EXIT TO LINK 4)
      N9   IMPACT MAPPING COMPLETE, CHECK FOR MORE INPUT
      N10  COMMENT OUTPUT REQUESTED

*V4  LA4   N6,N12
      N6   NO SORTING NECESSARY, GO TO DATA FITTING LINK
      N12  EXIT TO LINK 12 FOR ERROR

*V5  LA5   N9,N12
           (NO EXIT TO LINK 6)
      N9   PREDICTS SORTED CHECK FOR MORE INPUT
      N12  EXIT TO LINK 12 FOR ERROR

*V6  LA6   N12
      N12  EXIT TO LINK 12 FOR ERROR

*V7  LA7   N9,N10,N12

      N9   RETURN FOR MORE INPUT, NO RESIDUALS, STATISTICS REQUESTED
      N10  EXIT HERE TO PRINT PARTIALS
      N12  ERROR EXIT

*V8  LA8   N9,N12
           (NO EXIT TO LINK 10)
      N9   RETURN FOR MORE INPUT
      N12  ERROR EXIT

*V10 LA10  N8,N9
           (NO EXIT TO LINK X)
      N8   AFTER PARTIALS PRINT RETURN TO LINK 8 FOR RESIDUALS
           AND STATISTICS
      N9   NO OTHER REQUESTS, RETURN TO INPUT LINK

*X   END
    
```

VIII. CHECK CASES

JPL TECHNICAL MEMORANDUM NO. 33-204

SOURCE PROGRAM LISTING

```

*N1 DDPX N9
*N5 LA9 X,N11,N1,N4
55164/1.DE16
*N12 LA12 K
*N13 SPACE K
SCFORF=2
* WANT N9,(MOBPH11,200,(MOTGR,BRROPT),18
* WANT N11,(TAPEX),5
* WANT N12,(XN1,INJX),1,(XN2-1,INJY),1
* ETC (XN2-2,INJZ),1,(XN3-3,INJX),1
* ETC (XN4-4,INJY),1,(XN5-5,INJZ),1
* ETC (EPOCH,INJT),1,(EPOCH-1,INJT+1),1
* ETC (DUT,GRAY-2),1,(AU,SCALE1),1
* ETC (RE,SCALE1),1,(PDR,HARMH+5),1
* ETC (OD,HARMH+4),1,(OH,HARMH+3),1
* ETC (OJ,HARMH+2),1,(GRAVE,GRAY),1
* ETC (GRAVM,GRAY+1),1,(GRAVS,GRAY+2),1
* ETC (GRAVV,GRAY+3),1,(GRAVR,GRAY+4),1
* ETC (GRAVJ,GRAY+6),1,(STEP,H),1
* ETC (TARGET,TARBCD),1,(AREN,RADDPF+3),1
* ETC (GMS,RADDPF+4),1,(FMSS,RADDPF+5),1
* ETC (FAD,GASOPT+9),1,(FBO,GASOPT+12)
* ETC (FCO,GASOPT+15),1,(FAL,GASOPT+8)
* ETC (FBL,GASOPT+11),1,(FCL,GASOPT+14)
* ETC (FAZ,GASOPT+7),1,(FBZ,GASOPT+10)
* ETC (FCZ,GASOPT+13),1,(JETON,GASOPT),1
* ETC (JETON,GASOPT+2),1,(JETON-1,GASOPT+3),1
* ETC (JETOFF,GASOPT+5),1,(JETOFF-1,GASOPT+6),1
* ETC (TFIN,SECP50),1,(TFIN-1,SECP50+1),1
* ETC (FMSS,GASOPT+16),1
GASOPT+1=(CANOPU)
INJBCO=(EARTH)
RADDPF=.103169
*N11 LA11 N4,N6,N12
* WANT N9,(C137),1
* WANT N13,(CLPT,TRADE+1),1,(CLPT+1,TRADE),1
* ETC (CLPBT,TRADE+2),1,(CLPBR,TRADE+3),1
* ETC (CLPBT,TRADE+4),1,(CLPBR,TRADE+5),1
* ETC (TAPEX),6
*N2 LA2 N12,X
* WANT N11,(EPOX),14
*N2A LA2A N9,N12
*N3 LA3 N9,N10
*N4 LA4 N6,N12
*N5 LA5 N9,N12
*N6 LA6 N12
*N7 LA7 N9,N10,N12
* WANT N9,(APRIDR-19,APRIDR-19),20
*N8 LA8 N9,N12
*N10 LA10 N8,N9
*x END
    
```

THERE WERE NO GLARING SOURCE DECK ERRORS.
 THE OBJECT STRING HAS 00516 OCTAL OR 334 DECIMAL WORDS.

```

0000 134501 PAGE HEADING (23
0000 13450 IMARS CHECK CASE - FIT SIMULATED DATA, THEN MAP)
0000 13450 EPOCH (01
0000 13450 061102015.0757000
0000 13450 GEODETRIC POSITION AND VELOCITY AT EPOCH (02
0000 13450 X=.56082222E4 Y=.21966726E4 Z=-.32403740E4
0000 13450 DX=-.18839477E1 DY=-.10978654E2 DZ=-.12280548E1
0000 13450 OTHER OPTIONS AND CONSTANTS (03
0000 13450 TARGET=(IMARS)
0000 13450 INFO=3624200
0000 13450 ESTIMATE THESE PARAMETERS (04
0000 13450 X Y Z DX DY DZ MM AU
0000 13450 COVARIANCE MATRIX OF ESTIMATED PARANTERS (10
0000 13450 DIAG=10.,10.,10.,.01.,.01.,.01.,.1E-17,25.
0000 13450 POINTING TIMES, SAMPLE RATE, COUNT TIME (17
0000 13450 POINT=641102016,0,641103016,0,3600.,60.
0000 13450 JOBSET=641102016,0,641103016,0,3600.,60.
0000 13450 DATA FILE SIGMA (21
0000 13450 HA(4)=.001 DEC(4)=.002 CC(4)=.001
0000 13450 HA(5)=.002 DEC(5)=.001 CC(5)=.002
0000 13450 STATISTICS AND RESIDUALS FOR THESE DATA TYPES (14
0000 13450 HA(4)=.01 DEC(4)=.01 CC(4)=.01
0000 13450 HA(5)=.01 DEC(5)=.01 CC(5)=.01
0000 13450 WEIGHTS BY DATA TYPE AND STATION (16
0000 13450 HA(4)=.5 DEC(4)=.5 CC(4)=.1
0000 13450 HA(5)=.5 DEC(5)=.5 CC(5)=.1
0000 13450 TRANSMITTER ID TABLE (26
0000 13450 4,2038516,,641102916,0,5,2038516.
0000 13450 DEFINE CONTROL (27
0000 13450 KEY(2) KEY(14) KEY(16)
0000 13450 END DATA (0
    
```

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INPUT COVARIANCE MATRIX OF ESTIMATED PARAMETERS ITERATION NUMBER 0

	X	Y	Z	DX	DY	DZ	HM	AU
X	.10000000 02	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00
Y	.00000000 00	.10000000 02	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00
Z	.00000000 00	.00000000 00	.10000000 02	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00
DX	.00000000 00	.00000000 00	.00000000 00	.99999998-02	.00000000 00	.00000000 00	.00000000 00	.00000000 00
DY	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.99999998-02	.00000000 00	.00000000 00	.00000000 00
DZ	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.99999998-02	.00000000 00	.00000000 00
HM	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.99999998 18	.00000000 00
AU	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.25000000 02

INPUT J MATRIX OF ESTIMATED PARAMETERS ITERATION NUMBER 0

	X	Y	Z	DX	DY	DZ	HM	AU
X	.99999999-01	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00
Y	.00000000 00	.99999999-01	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00
Z	.00000000 00	.00000000 00	.99999999-01	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00
DX	.00000000 00	.00000000 00	.00000000 00	.10000000 03	.00000000 00	.00000000 00	.00000000 00	.00000000 00
DY	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.10000000 03	.00000000 00	.00000000 00	.00000000 00
DZ	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.10000000 03	.00000000 00	.00000000 00
HM	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.99999998 18	.00000000 00
AU	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.00000000 00	.99999999-01

CASE 1 IBSYS-JPTRAJ-SPACE 022265

DOUBLE PRECISION EPHEMERIS TAPE - EPHEMI

GME	.39860663 06	J	.16234500-02	H	-.57499999-05	D	.78749999-05	RE	.63781650 04	REM	.63783112 04
G	.66709998-19	A	.88781796 29	B	.80800194 29	C	.88369716 29	GME	.41780741-02	AU	.14959850 09
GMM	.49026293 04	GMS	.13271611 12	GMV	.32476627 04	GMA	.42977368 05	GMC	.37918700 08	GMD	.12670935 09
EGM	.39860320 06	MGM	.49027779 04	JA	.29200000-02	HA	.00000000 00	DA	.00000000 00	RA	.34170000 04

RADIATION PRESSURE INPUT

ARA	.11120000 02	GB	.95999999-01	MAS	.25900000 03	G81	.00000000 00	G82	.00000000 00	SC	.10310000 09
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INJECTION CONDITIONS MARS 235700551423202200000000 J.D.= 2438728.13052083 NOV. 28,1964 15 07 57.000

GEOCENTRIC XO = 56682222 04 YO = -21466726 04 ZO = -32403748 04 DX0 = -18839476 01 DY0 = 10978654 02 DZ0 = -12280548 01

CARTESIAN TO 54476999 05 GHA = -29452970 03 GHO = -66920765 02

DATE OF RUN 022265A 13660 EARTH IS THE CENTRAL BODY FOR INTEGRATION COWELL EQUATIONS OF MOTION

PROBE IS IN EARTH'S SHADOW

0 DAYS 0 HRS. 0 MIN. 0.000 SEC. 235700551423202200000000 J.D.= 2438728.13052083 NOV. 28,1964 15 07 57.000

GEOCENTRIC EQUATORIAL COORDINATES

X	.56682219 04	Y	-.21466724 04	Z	-.32403747 04	DX	-.18839476 01	DY	.10978653 02	DZ	-.12280547 01
R	.68729157 04	DEC	-.28129771 02	RA	.20742711 02	V	-.11206614 02	PTH	.12650650 02	AZ	.90421379 02
R	.68729154 04	LAT	-.28129772 02	LOM	.86213015 02	VE	.10775808 02	PTE	.13165318 02	AZE	.90439129 02
XS	-.58995549 08	YS	-.12408705 09	ZS	-.53810146 08	DXS	.27792202 02	DYS	-.10840364 02	DZS	-.47011071 01
XM	-.38523466 06	YM	-.44736395 05	ZM	.18645579 05	DXM	.37058146-01	DYM	-.92038371 00	DZM	-.41497853 00
XT	-.19750066 09	YT	.58811766 08	ZT	.33854926 08	DXT	.87298950 01	DYT	-.21590059 02	DZT	-.91209471 01
RS	-.14755881 09	VS	.30199675 02	RH	.30827148 06	VH	.10102904 01	RT	.20883364 09	VT	.25010665 02
QED	-.28291871 02	ALT	.49950940 03	LOS	.31004216 03	RAS	-.24457186 03	RAM	.18662395 03	LOM	.25209425 03
QUT	-.35000000 02	DT	.37500000 01	DR	-.24543145 01	SHA	.62358861 04	DES	-.21387223 02	DEM	.27525167 01
CCL	.11901304 03	MCL	-.21144471 03	ICL	-.21574659 03						

GEOCENTRIC CONIC

235700551326202356642400 J.D.= 2438728.12770659 NOV. 28,1964 15 03 53.850

SMA	-.41536075 05	ECC	.11580720 01	B	.24259727 05	SLR	.14169237 05	APD	.00000000 00	RCA	.65656922 04
VH	.30978206 01	C3	.95964924 01	G1	.75152290 05	TFF	.24314986 03	TF	-.28142345-02	LTF	-.25588951-01
TA	.23551647 02	MTA	.14971233 03	EA	.64722201 01	MA	.10390285 01			TFI	.00000000 00

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE

X	.56682219 04	Y	-.21466724 04	Z	-.32403747 04	DX	-.18839476 01	DY	.10978653 02	DZ	-.12280547 01
IHC	.28132666 02	LAN	-.11163641 03	APF	.24556020 03	MX	-.35740510 00	MY	.93392694 00	MZ	-.64857359-02
WX	.43829295 00	WY	.17385475 00	WZ	.08185818 00	PX	.89883014 00	PY	-.86854062-01	PZ	-.42920527 00
QX	.19040406-02	QY	.98093365 00	QZ	-.19433341 00	RX	-.21994247 00	RY	.16164777 00	RZ	-.96202662 00
BX	.45496172 00	BY	.80323618 00	BZ	-.38447559 00	TX	.59221295 00	TY	.80578151 00	TZ	.00000000 00
SX1	-.77710383 00	SY1	.41972693 00	SZ1	-.46897637 00	DA1	-.27967872 02	RA1	.28374231 02		
SX0	-.77518326 00	SY0	.56972462 00	SZ0	.27295548 00	DA0	.15840210 02	RA0	.14368579 03		

B1Q .24222576 05 BRQ .13420740 04 B .24259727 05 THA .31712828 01 T VECTOR IN EARTH EQUATOR PLANE

ALL VECTORS REFERENCED TO ECLIPTIC PLANE

X	.56682219 04	Y	.68027003 03	Z	-.38269391 04	DX	-.18839476 01	DY	.95837711 01	DZ	-.54945680 01
IHC	.42277832 02	LAM	.13934388 03	APF	-.21231137 03	MX	-.35740509 00	MY	.85425011 00	MZ	-.37751597 00
WX	.43829294 00	WY	.51035282 00	WZ	.79880141 00	PX	.89883011 00	PY	-.25060393 00	PZ	-.35958584 00
QX	.19040406-02	QY	.82264062 00	QZ	-.56855842 00	RX	-.18420546-01	RY	.15001224-01	RZ	-.99971770 00
8X	.45496171 00	8Y	.58396360 00	8Z	-.67230674 00	TX	.63146777 00	TY	.77540212 00	TZ	.00000000 00

JPL TECHNICAL MEMORANDUM NO. 33-204

CASE 1

IBSYS-JPIRAJ-SPACE 022265

SX1 .77710380 00 SY1 .19849447 00 SZ1 -.59725159 00 DAI -.36673309 02 RA1 .14328600 02
 SX0 -.77518323 00 SY0 .63128952 00 SZ0 -.23756120-01 DAO .13612515 01 RAO .14084150 03
 BTC .22379497 05 BRC -.93644235 04 B .24259727 05 THA .33729373 03 T VECTOR IN ECLIPTIC PLANE
 HELIOCENTRIC ECLIPTIC COORDINATES
 X .59001217 08 Y .13525276 09 Z -.41570000 04 DX -.29676149 02 DY .21399603 02 DZ -.56944124 01
 A .14756169 09 LAT -.16140947-02 LON .66431758 02 V .36997370 02 PTH .12089898 02 AZ .98735356 02
 XE .58995549 08 YE .13525209 09 ZE -.33000000 03 DXE -.27792202 02 DYE .11815832 02 DZE .15556812-03
 XT -.13850512 09 YT .20267817 09 ZT -.76614060 07 DXT -.19052307 02 DYT -.11620746 02 DZT .22182012 00
 LTE -.12813608-03 LOE .66433669 02 LTT .17875885 01 LDT .12434771 03 RST .24560294 09 VST .22326273 02
 EPS .65134214 02 ESP .13988227-01 SEP .11486336 03 EPH .28249721 02 EMP .48000350 00 MEP .15127023 03
 MPS .61567036 02 MSP .13507880 00 SHP .11829015 03 SEM .61571418 02 EMS .11829583 03 ESM .13270400 00
 EPT .39785299 02 ETP .27453512-18 TEP .14021349 03 TSP .85283946 02 TYP .57933486 02 STP .36782563 02
 SET .85286588 02 STE .36781888 02 EST .57931519 02 RPM .39431215 06 RPT .20883893 09 SPN .-29894790 01
 SAC .22280601-09
 GCE .24098695 03 GCT .27673355 03 SIP .85283019 02 CPT .34329569 02 SIN .84328642 02
 REP .68729157 04 VFP .11206614 02 CPE .12085570 03 CPS .10111911 03
 0 DAYS 0 HRS. 9 MIN. 49.244 SEC. 235700551646202437164374 J.D.= 2438728.13734078 NOV. 28,1964 15 17 46.244
 PROBE IS LEAVING EARTH'S SHADOW
 2 DAYS 0 HRS. 52 MIN. 3.000 SEC. 235700677340202000000000 J.D.= 2438730.16666666 NOV. 30,1964 16 00 00.000
 0 DAYS 0 HRS. 52 MIN. 18.000 SEC. 23570067734320260000000000 J.D.= 2438730.16684028 NOV. 30,1964 16 00 15.000
 215542322602 214412016141 614626031437 601731456722 204537405524 601471557132 EARTH
 21149808915 17481059072 000000000000 INITIAL
 623733473151 223571732535 222507660544 602504407717 201741457741 200715435157 EARTH
 235700677343 202600000000 235700677343 202600000000 END

J MATRIX ITERATION NUMBER 1

	X	Y	Z	DX	DY	DZ	MM	AU
X	.86586481 07	.34045452 07	-.50691355 07	-.21792881 10	.13770335 11	-.15016493 10	.23125263 07	-.77185333-04
Y	.34045452 07	.13396596 07	-.19929949 07	-.85587537 09	.54135596 10	-.62224241 09	.91247442 06	-.30475400-04
Z	-.50691355 07	-.19929949 07	.29680390 07	.12758964 10	-.80622930 10	-.92586914 09	-.13449765 07	.45040778-04
DX	-.21792881 10	-.85587537 09	-.12758964 10	.54958336 12	-.34665872 13	.39774309 12	-.58092013 09	.19371962-01
DY	.13770335 11	.54135596 10	.80622930 10	-.34665872 13	.21901237 14	-.25151332 13	.36716555 10	-.12253934 00
DZ	-.15016493 10	-.62224241 09	.92586914 09	.39774309 12	-.25151332 13	.28909285 12	-.42327558 09	.14149069-01
MM	.23125263 07	.91247442 06	-.13449765 07	-.58092013 09	.36716555 10	-.42327558 09	.99999998 18	-.29593108-04
AU	-.77185333-04	-.30475400-04	.45040778-04	-.19371962-01	-.12253934 00	.14149069-01	-.29593198-04	.39999999-01

CORRELATIONS BASED ON J MATRIX ITERATION NUMBER 1

	X	Y	Z	DX	DY	DZ	MM	AU
X	-.10000000 01	-.99962436 00	.99994028 00	-.99901687 00	-.99996582 00	.99969255 00	-.78588977-06	.13115346-06
Y	-.99962436 00	-.10000000 01	.99948009 00	-.99746235 00	-.99942768 00	.99987035 00	-.78835797-06	.13116504-06
Z	.99994028 00	.99948009 00	-.99999999 00	-.99899622 00	.99997574 00	-.99953067 00	.78359453-06	-.13071971-06
DX	.99901687 00	.99746235 00	-.99899622 00	-.10000000 01	.99919678 00	-.99785440 00	.78360940-06	-.13065524-06
DY	-.99996582 00	-.99942768 00	.99997574 00	.99919678 00	-.10000000 01	.99955799 00	-.78456262-06	.13092157-06
DZ	.99969255 00	.99948009 00	-.99953067 00	-.99785440 00	.99955799 00	-.10000000 01	.78723525-06	-.13157676-06
MM	-.78588977-06	-.78835797-06	.78359453-06	.78360940-06	-.78456262-06	.78723525-06	-.10000000 01	-.14796599-12
AU	.13115346-06	.13116504-06	-.13071971-06	-.13065524-06	.13092157-06	-.13157676-06	.14796599-12	-.10000000 01

MARS CHECK CASE - FIT SIMULATED DATA, THEN MAP

ITERATION NUMBER	1	EPOCH 64/11/28 150757.000	CLOCK 135150	SOS .20501 00	QSOS .20501 00		
Q	DD	STDEVQ	APRIORI STDEVQ	OLD Q	NEW Q	NOMINAL Q	DOINGM
X	-.11623597-01	.75099736 00	.31622776 01	-.56682222 04	.56682105 04	-.56682222 04	-.11596680-01
Y	.13752168-01	.25575302 00	.31622776 01	-.21465726 04	.21466863 04	-.21466726 04	-.13763428-01
Z	-.57554049-02	.77465767 00	.31622776 01	-.32403748 04	-.32403805 04	-.32403748 04	-.57678223-02
DX	-.15225409-04	.57259488-03	.99999999-01	-.18839477 01	-.18839629 01	-.18839477 01	-.15228987-04
DY	-.74521106-06	.78780906-03	.99999999-01	-.10978654 02	.10978653 02	-.10978654 02	-.71525574-06
DZ	-.12002572-05	.75434680-03	.99999999-01	-.12280548 01	-.12280560 01	-.12280548 01	-.12069941-05
MM	-.34015613-17	.99999997-09	.99999997-09	-.32383419-06	.32383419-06	-.32383419-06	-.35527136-14
AU	.28060658-08	.50000000 01	.50000000 01	-.14959850 09	.14959850 09	-.14959850 09	-.00000000 00

JPL TECHNICAL MEMORANDUM NO. 33-204

J INVERSE ITERATION NUMBER 1. Table with columns X, Y, Z, DX, DY, DZ, HM, AU and rows X, Y, Z, DX, DY, DZ, HM, AU.

COVARIANCE MATRIX OF ESTIMATED PARAMETERS ITERATION NUMBER 1. Table with columns X, Y, Z, DX, DY, DZ, HM, AU and rows X, Y, Z, DX, DY, DZ, HM, AU.

CORRELATION MATRIX OF ESTIMATED PARAMETERS ITERATION NUMBER 1. Table with columns X, Y, Z, DX, DY, DZ, HM, AU and rows X, Y, Z, DX, DY, DZ, HM, AU.

STATION NUMBER 41 64/11/28 ITERATION NUMBER 1 PASS NUMBER 000000 PAGE 1
FREQUENCY Z038516.0

Table with columns TIME, TC, Q, DEC, HA, CC3 and rows 160000, 170000, 180000, 190000, 200000, 210000, 220000, 230000.

64/11/29

Table with columns STATION NUMBER, DEC, HA, CC3 and rows 000000, 150000, 160000.

FREQUENCY .0

Table with columns STATION NUMBER, DEC, HA, CC3 and rows 170000, 180000, 190000, 200000, 210000.

FREQUENCY Z038516.0

Table with columns STATION NUMBER, DEC, HA, CC3 and rows 220000, 230000.

64/11/30

Table with columns STATION NUMBER, DEC, HA, CC3 and rows 000000, 010000.

FREQUENCY .0

Table with columns STATION NUMBER, DEC, HA, CC3 and rows 150000, 160000.

JPL TECHNICAL MEMORANDUM NO. 33-204

MARINER STATISTICS			STATION 41				ITERATION 1	
PASS	DATA TYPE	BEGINNING TIME	END TIME	NUMBER OF POINTS	STD DEV	RMS	FIRST MOMENT	SECOND MOMENT
000000	CC3	11/28-160000	11/29-160000	11	.574-03	.589-03	.133-03	.347-06
	HA	11/28-160000	11/30-160000	22	.101-02	.101-02	-.461-04	.103-05
	DEC	11/28-160000	11/30-160000	22	.158-02	.165-02	-.459-03	.271-05

STATION NUMBER 51 64/11/28 ITERATION NUMBER 1 PASS NUMBER 000000 PAGE 2

FREQUENCY2038516.0

TIME	TC	Q	DEC	HA	CC3			
220000	60	41	.13129207 02 .735-01	.0017	-.28861228 03 .754-01	.0004		
230000	60	41	.13532427 02 .667-01	.0005	-.30329097 03 .686-01	.0046		
64/11/29								
000000	60	41	.13838208 02 .654-01	-.0009	-.31818084 03 .674-01	-.0044		
FREQUENCY .0								
010000	60	0	.14068252 02 .652-01	.0010	-.33319524 03 .671-01	.0026		
020000	60	0	.14235436 02 .649-01	-.0008	-.34827475 03 .671-01	.0005		
030000	60	0	.14349870 02 .649-01	.0002	-.33738108 01 .671-01	.0035		
040000	60	0	.14420789 02 .652-01	.0005	-.18457121 02 .671-01	-.0048		
050000	60	0	.14456847 02 .652-01	.0009	-.33498143 02 .674-01	.0005		
060000	60	0	.14465484 02 .659-01	-.0010	-.48477208 02 .681-01	.0005		
070000	60	0	.14461425 02 .684-01	.0015	-.63393088 02 .706-01	.0007		
080000	60	0	.14359558 02 .112 00	-.0004	-.78067596 02 .116 00	.0057		
FREQUENCY2038516.0								
220000	60	51	.14953242 02 .857-01	-.0002	-.28509123 03 .886-01	-.0014	.14302250 05 .172-01	.0009
230000	60	51	.15068582 02 .676-01	.0000	-.30005508 03 .698-01	.0017	-.14459947 05 .135-01	.0001
64/11/30								
000000	60	51	.15137918 02 .657-01	-.0019	-.31513791 03 .681-01	-.0009	.14735132 05 .132-01	-.0002
010000	60	51	.15189322 02 .652-01	-.0002	-.33026529 03 .676-01	-.0031	-.15106839 05 .131-01	.0021
020000	60	51	.15225421 02 .649-01	.0009	-.34541624 03 .674-01	.0017	.15546990 05 .130-01	-.0020
030000	60	51	.15246021 02 .649-01	.0002	-.57495905 00 .674-01.	.0012	.16022491 05 .130-01	.0007
040000	60	51	.15254266 02 .652-01	.0004	-.15727202 02 .674-01	.0012	.16497762 05 .130-01	-.0017
050000	60	51	.15248872 02 .652-01	.0009	-.30859974 02 .676-01	-.0003	.16937468 05 .131-01	.0029
060000	60	51	.15231617 02 .659-01	-.0003	-.45960994 02 .681-01	.0005	.17309181 05 .132-01	.0002
070000	60	51	.15206400 02 .679-01	-.0001	-.61013895 02 .703-01	-.0045	-.17585710 05 .135-01	-.0012
080000	60	51	.15120052 02 .925-01	-.0006	-.75933573 02 .959-01	-.0002	.17746981 05 .185-01	.0010

MARINER STATISTICS			STATION 51				ITERATION 1	
PASS	DATA TYPE	BEGINNING TIME	END TIME	NUMBER OF POINTS	STD DEV	RMS	FIRST MOMENT	SECOND MOMENT
000000	CC3	11/29-220000	11/30-080000	11	.144-02	.146-02	.255-03	.213-05
	HA	11/28-220000	11/30-080000	22	.267-02	.269-02	.334-03	.722-05
	DEC	11/28-220000	11/30-080000	22	.794-03	.850-03	.303-03	.723-06
0000	135231	OFFLINE CONTROL				127		
0000	13523	KEY(5)						
0000	13523	END DATA				ID		

JPL TECHNICAL MEMORANDUM NO. 33-204

CASE 1

IBSYS-JPTRAJ-SPACE 022265

DOUBLE PRECISION EPHEMERIS TAPE - EPHEMI

GME .39800063 06 J .16234500-02 H -.57499999-05 D .78749999-05 RE .63781650 04 REM .63783112 04
 G .66709998-19 A .88781796 29 B .88800194 29 C .88836976 29 DHE .41780741-02 AU .14959850 09
 GMN .49026293 04 GMS .13271411 12 GNV .32476627 06 GHA .42977367 05 GMC .37918700 08 GMJ .12670935 09
 EGM .39800063 06 GGN .49027779 04 JA .29200000-02 HA .00000000 00 DAK .00000000 00 RA .34170000 04
 RADIATION PRESSURE INPUT
 ARA .11120000 02 GB .95999999-01 MAS .25900000 03 G81 .00000000 00 G82 .00000000 00 SC .10310000 09

INJECTION CONDITIONS MARS 23570055142320220000000 J.D.= 2438728.13052083 NOV. 28,1964 15 07 57.000

GEOCENTRIC X0 .56682105 04 Y0 .21466863 04 Z0 -.32403805 04 DX0 .18839629 01 DY0 .10978653 02 DZ0 .12280559 01
 CARTESIAN T0 .54476999 05 GHA .29452970 03 GHD .66920765 02
 DATE OF RUN 022265A 13525 EARTH IS THE CENTRAL BODY FOR INTEGRATION COMELL EQUATIONS OF MOTION

PROBE IS IN EARTH'S SHADOW

0 DAYS 0 HRS. 0 MIN. 0.000 SEC. 23570055142320220000000 J.D.= 2438728.13052083 NOV. 28,1964 15 07 57.000

GEOCENTRIC EQUATORIAL COORDINATES
 X .56682103 04 Y .21466862 04 Z -.32403804 04 DX .18839629 01 DY .10978653 02 DZ .12280559 01
 R .68729131 04 DEC .-28129837 02 RA .20742872 02 V .11206616 02 PTH .12650726 02 AZ .90421329 02
 XS .-58995549 08 YS .-12408705 09 ZS .-53810146 08 OXS .27792202 02 VES .10775811 02 PTE .13165397 02 AZE .90439077 02
 XM .-38523466 06 YM .-44736395 05 ZM .18645579 05 OXM .37058146-01 DYH .92038371 00 DZH .-41497853 00
 XT .-19750066 09 YT .58811766 08 ZT .33854926 08 DXT .87288950 01 DYT .-21590059 02 DZT .-91209471 01
 RS .-14755881 09 VS .30199675 02 RM .38827148 06 VM .10102904 01 RT .20885304 09 VT .25010665 02
 GEO .-28291938 02 ALT .49950689 03 LOS .31004216 03 RAS .24451186 03 RAH .18662395 03 LDH .25209425 03
 DUT .35000000 02 DT .-37500000 01 DR .24542394 01 SHA .62358811 04 DES .-21387223 02 DEM .27525167 01
 CCL .11901320 03 NCL .21144487 03 TCL .21574675 03

GEOCENTRIC CONIC

EPOCH OF PERICENTER PASSAGE 23570055132620235632500 J.D.= 2438728.12770658 NOV. 28,1964 15 03 53.849
 SMA .-41536060 05 ECC .11580719 01 B .24259711 05 SLR .14169223 05 APO .00000000 00 RCA .65658662 04
 VH .30978211 01 C3 .-95864959 01 CL .75152253 05 TFP .24315096 03 TF .-28142471-02 LTF .-25588937-01
 TA .23551788 02 MTA .14971234 03 EA .64722590 01 MA .10390338 01 TF1 .00000000 00

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
 X .56682103 04 Y .21466862 04 Z -.32403804 04 DX .18839629 01 DY .10978653 02 DZ .12280559 01
 INC .28132733 02 LAN .11163647 03 APF .24566016 03 WK .-35740735 00 PY .-86854203-01 PZ .-42960603 00
 MX .43829371 00 MY .-17389584 00 WZ .-88185763 00 PX .89882977 00 RY .16164795 00 RZ .-96202654 00
 QX .19036569-02 QY .98093352 00 QZ .-19433418 00 RX .-21994273 00 TY .80578150 00 TZ .00000000 00
 BX .45496110 00 BY .80323605 00 BZ .-38447660 00 TX .-59221295 00 TY .80578150 00 TZ .00000000 00
 SXI .77710339 00 SYI .-41972664 00 SZI .-46897743 00 DAI .-27967940 02 RAI .28374228 02
 SXD .-77518320 00 SYD .56972458 00 SZD .-27295579 00 DAD .15840229 02 RAD .14368579 03

ALL VECTORS REFERENCED TO ECLIPTIC PLANE
 X .56682103 04 Y .68028036 03 Z .-38269499 04 DX .18839629 01 DY .95837699 01 DZ .-54945689 01
 INC .42277902 02 LAN .13934386 03 APF .21231139 03 MX .-35740735 00 MY .85424963 00 NZ .-37751494 00
 WX .43829370 00 WY .51035333 00 WZ .-73989060 00 PX .89882980 00 PY .25000437 00 PZ .-35958650 00
 QX .19036569-02 QY .82264020 00 QZ .-68559110 00 RX .-18420855-01 RY .15001476-01 RZ .-99371775 00
 OX .45496101 00 OY .58376315 00 OZ .-67230762 00 TX .63146777 00 TY .77540213 00 TZ .00000000 00

CASE 1

IBSYS-JPTRAJ-SPACE 022265

2

SXI .77710346 00 SYI .19849367 00 SZI .-59725243 00 DAI .-36673369 02 RAI .14328550 02
 SXD .-77518328 00 SYD .63128954 00 SZD .-23756519-01 DAD .-13612746 01 RAD .14084150 03
 BTC .22379481 05 BRC .-93644400 04 B .24259711 05 THA .33729368 03 I VECTOR IN ECLIPTIC PLANE

HELIOCENTRIC

ECLIPTIC COORDINATES

X .59901217 08 Y .13525276 09 Z .-41570000 04 DX .-29676165 02 DY .21399602 02 DZ .-54944132 01
 R .14750169 09 LAI .-16140947-02 LON .66643178 02 V .36997382 02 PTH .12089884 02 AZ .98735355 02
 XE .58995549 08 YE .13525209 09 ZE .-33000000 03 DXE .-27792202 02 DYE .11815832 02 DZE .15556812-03
 XT .-13850512 09 YT .20267817 09 ZT .76614060 07 DXT .-19062307 02 DYT .-11620746 02 DZT .22182012 00
 LTE .-12813608-03 LDE .66433669 02 LTI .17875885 01 LDT .12434771 03 RST .24560294 09 VST .22328273 02
 EPS .65134159 02 ESP .13988227-01 SEP .11486342 03 EPH .28249846 02 EMP .47990158 00 MEP .15127010 03
 MPS .61567035 02 MPD .13507800 00 SXP .11829816 03 SEM .61571418 02 EHS .11829583 03 ESM .13270400 00
 EPT .39785456 02 ETP .27453512-18 TEP .14021333 03 TPS .85283946 02 TSP .-57933486 02 STP .36782563 02
 SEI .85286588 02 STE .36781888 02 EST .57931519 02 RPM .39431214 06 RPT .20883893 09 SPN .-29895868 01
 SAC .22280601-09
 GCE .24098679 03 GCT .27673355 03 SIP .85283019 02 CPT .84329569 02 SIN .84328642 02
 REP .68729131 04 VEP .11206616 02 CPE .12085585 03 CPS .10111911 03

0 DAYS 0 HRS. 9 MIN. 49.241 SEC. 235700551646202436644705 J.D.= 2438728.13734075 NOV. 28,1964 15 17 46.241
 PROBE IS LEAVING EARTH'S SHADOW

8 DAYS 19 HRS. 24 MIN. 2.979 SEC. 235701335107202775352514 J.D.= 2438736.93888865 DEC. 7,1964 10 31 59.980

8 DAYS 19 HRS. 24 MIN. 2.979 SEC. 235701335107202775352514 J.D.= 2438736.93888865 DEC. 7,1964 10 31 59.980
 CHANGE OF PHASE OCCURS AT THIS POINT SUN IS THE CENTRAL BODY FOR INTEGRATION COMELL EQUATIONS OF MOTION

225 DAYS 2 HRS. 44 MIN. 31.060 SEC. 235723202427202007527522 J.D.= 2438953.24476920 JULY 11,1965 17 52 28.060

GEOCENTRIC

EQUATORIAL COORDINATES

X .-20952635 09 Y .-22337826 08 Z .-94661789 07 DX .-13367881 02 DY .-23273616 02 DZ .-10141585 02
 R .21092624 09 DEC .-25722458 01 RA .18608537 03 V .28691692 02 PTH .34373873 02 AZ .11345038 03
 XS .-21092624 09 YS .-25722458 01 LON .34854236 03 VE .15343772 05 PTE .60490716-01 AZE .26996480 03
 XM .-50004426 08 YM .13176278 09 ZM .57140097 08 OXM .-27634650 02 DYS .-88887630 01 OZS .-38558931 01
 XN .-16125000 05 YN .36611200 06 ZN .-16854550 06 DXN .97207475 00 DYN .-20357370-01 DZN .10283720 00
 XT .-21150188 09 YT .-22632994 08 ZT .-93652949 07 DXT .-92234842 01 DYT .-22096207 02 DZT .10412922 02
 RS .15207919 09 VS .29283897 02 RM .40336781 06 VM .97271122 00 RT .21291549 09 VT .26110227 02
 GEO .-25897513 01 ALT .21091986 09 LOS .27323891 03 RAS .11078193 03 RAM .26747809 03 LOM .69935081 02
 DUT .35000000 02 DT .86399999 05 DR .16199062 02 SHA .-20585269 09 DES .22069798 02 DEM .-24698652 02
 CCL .10206468 03 NCL .18000385 03 TCL .92811081 01

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HELIOCENTRIC
 X -15952193 09 Y -16787898 09 Z -20262100 06 DX -14266770 02 DY -15698153 02 DZ -43642342-01
 R -23158289 09 LAT -50131026-01 LON -22646218 03 V -21212605 02 PTH -41969292 01 AZ -90121874 02
 XE -50004426 08 YE -14361901 09 ZE -28500000 02 OXE -27634650 02 OYE -96890669 01 OZE -11010766-02
 XT -16149745 09 YT -16810964 09 ZT -41261150 06 DXT -18411166 02 DYT -14725897 02 DZT -76102304 00
 LYE -10737647-04 LDE -28919679 03 LTT -10141465 00 LDT -22614924 03 RST -23311467 09 VST -23588180 02
 EPS -39857595 02 ESP -62734623 02 SEP -77407780 02 EPM -10835236 00 EMP -98784596 02 MEP -81107116 02
 MPS -39965850 02 MSP -62679107 02 SHP -77355039 02 SEM -15847546 03 EMS -21468911 02 ESM -55952909-01
 EPT -17402787 03 ETP -59161249 01 TEP -54625775-01 TSP -13982710 03 TSP -31712556 00 STP -39855791 02
 SET -77408312 02 STE -39544090 02 EST -63047593 02 RPM -21086426 09 RPT -19999992 07 SPN -39855862 02
 SAC -90460971-10
 GCE -25793531 03 GCT -87216419 02 SGP -13973033 03 SPT -93937702 02 SIN -93840928 02
 REP -21092624 09 VEP -28691692 02 CPE -91869014 02 CPS -82500247 02

HELIOCENTRIC CONIC
 EPOCH OF PERICENTER PASSAGE 2357202470660202547527522 J.D.= 2438723.37336585 NOV. 23, 1964 20 57 38.810
 SMA -19063377 09 ECC -22638500 00 B -18568449 09 SLR -18086375 09 APO -23379039 09 RCA -14747714 09
 VA -20955964 02 C3 -69617315 03 CI -48993033 10 TFP -19860889 08 TF -47571538 01 PER -52562087 03
 TH -16533578 03 HTA -18000000 03 EA -16159514 03 MA -15749984 03

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
 X -15952193 09 Y -15410060 09 Z -66606276 08 DX -14266770 02 DY -14384853 02 DZ -62856916 01
 INC -23492319 02 LAN -30826009 00 APF -60843575 02 MX -72491694 00 MY -63112116 00 MZ -27600995 00
 NX -21446739-02 NY -39862036 00 WZ -91711352 00 PX -48287950 00 PY -80351821 00 PZ -34811732 00
 OX -87568406 00 OY -44210871 00 QZ -19420888 00 RX -17931470 00 RY -29838215 00 RZ -93745086 00
 QX -87568416 00 QY -44210876 00 QZ -19420890 00 TX -85713101 00 TY -51509847 00 TZ -00000000 00
 DAP -20372204 02 RAP -96999964 02

BTO -18165617 09 BRO -38467694 08 B -18568449 09 THA -11956381 02 T VECTOR IN EARTH EQUATOR PLANE

ALL VECTORS REFERENCED TO ECLIPTIC PLANE
 X -15952193 09 Y -16787898 09 Z -20262100 06 DX -14266770 02 DY -15698153 02 DZ -43542342-01
 INC -13177922 00 LAN -68820911 02 APF -35230551 03 MX -72491693 00 MY -68883299 00 MZ -21271140-02
 NX -21446737-02 NY -83096182-03 WZ -99999736 00 PX -48287949 00 PY -87586647 00 PZ -30799250-03
 OX -87568391 00 OY -48287487 00 QZ -22793191-02 RX -14870399-03 RY -26966991-03 RZ -99999969 00
 BX -87568413 00 BY -48287899 00 BZ -22793196-02 TZ -87568673 00 TY -48287963 00 TZ -00000000 00
 DAP -17645793-01 RAP -61126355 02

BTC -18568398 09 BRC -42323416 06 B -18568446 09 THA -13059492 00 T VECTOR IN ECLIPTIC PLANE

ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET
 X -17224305 09 Y -15479968 09 Z -16642050 06 DX -15289946 02 DY -14689711 02 DZ -63605690 00
 INC -17264500 01 LAN -31941926 03 APF -19329738 03 MX -66815475 00 MY -74341242 00 MZ -30118211-01
 NX -19598038-01 NY -22880996-01 WZ -99994606 00 PX -88868259 00 PY -45847028 00 PZ -69293238-02
 OX -45810362 00 OY -88841500 00 QZ -29319061-01 RX -61581179-02 RY -31769557-02 RZ -99997587 00
 BX -45810368 00 BY -88841512 00 BZ -29319065-01 TZ -45848135 00 TY -88870403 00 TZ -00000000 00
 DAP -39702438 00 RAP -15271084 03

BTO -18560465 09 BRO -54442254 07 B -18568448 09 THA -35831986 03 T VECTOR IN ORBIT PLANE OF TARGET

225 DAYS 2 HRS. 44 MIN. 31.060 SEC. 235723202427202007527522 J.D.= 2438953.24476920 JULY 11, 1965 17 52 28.060
 CHANGE OF PHASE OCCURS AT THIS POINT MARS IS THE CENTRAL BODY FOR INTEGRATION COLL EQUATIONS OF MOTION

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225 DAYS 2 HRS. 44 MIN. 31.060 SEC. 235723202427202007527522 J.D.= 2438953.24476920 JULY 11, 1965 17 52 28.060

GEOCENTRIC
 X -20952635 09 Y -22337827 08 Z -94661791 07 DX -13367881 02 DY -23273616 02 DZ -10141565 02
 R -21092624 09 DEC -25722458 01 RA -18608537 03 V -28691692 02 PTH -34732873 02 AZ -11345038 03
 R -21092624 09 LAT -25722458 01 LON -34856235 03 VE -15343772 05 PTE -60490716-01 AZE -26996480 03
 XE -50004427 08 YE -14361901 09 ZE -28500000 02 OXE -27634650 02 OYE -96890669 01 OZE -11010766-02
 XM -16149745 09 YM -16810964 09 ZM -41261150 06 DXM -18411166 02 DYM -14725897 02 DZM -76102304 00
 XT -21150188 09 YT -22623995 08 ZT -93652949 07 DXT -92234842 01 DYT -22096207 02 DZT -10412222 02
 RS -15207519 09 VS -29283987 02 RM -40336780 06 VM -97771098 00 RT -21291549 09 VT -26110222 02
 GED -25897513 01 ALT -71091986 09 LDS -27323891 03 RAS -11078193 03 RAS -26747793 03 LDM -69934925 02
 DUT -35800000 02 DT -38400000 04 DCR -16199062 02 SHA -20585269 09 DES -22069798 02 DEN -24686832 02
 CCL -10206468 03 MCL -18000385 03 IRL -92811038 01

HELIOCENTRIC
 X -15952193 09 Y -16787898 09 Z -20262100 06 DX -14266770 02 DY -15698153 02 DZ -43642342-01
 R -23158289 09 LAT -50131026-01 LON -22646218 03 V -21212605 02 PTH -41969292 01 AZ -90121874 02
 XE -50004427 08 YE -14361901 09 ZE -28500000 02 OXE -27634650 02 OYE -96890669 01 OZE -11010766-02
 XT -16149745 09 YT -16810964 09 ZT -41261150 06 DXT -18411166 02 DYT -14725897 02 DZT -76102304 00
 LYE -10360887-04 LDE -28919679 03 LTT -10141465 00 LDT -22614924 03 RST -23311467 09 VST -23588180 02
 EPS -39857594 02 ESP -62734623 02 SEP -77407780 02 EPM -10812639 00 EMP -98784597 02 MEP -81107116 02
 MPS -39965851 02 MSP -62679107 02 SHP -77355038 02 SEM -15847546 03 EMS -21468911 02 ESM -55952909-01
 EPT -17402788 03 ETP -59161291 01 TEP -54625775-01 TSP -13982710 03 TSP -31712556 00 STP -39855791 02
 SET -77408312 02 STE -39544090 02 EST -63047593 02 RPM -21086426 09 RPT -19999992 07 SPN -39855862 02
 SAC -90460971-10
 GCE -25793531 03 GCT -87216416 02 SGP -13973032 03 SPT -93937700 02 SIN -93840926 02
 REP -21092624 09 VEP -28691692 02 CPE -91869014 02 CPS -82500245 02

AREOCENTRIC
 X -18755248 07 Y -23066303 06 Z -20999099 06 DX -41443964 01 DY -97225593 00 DZ -71738064 00
 R -20000000 07 DEC -60269085 01 RA -66597225 01 V -40316936 01 PTH -82615601 02 AZ -29904489 03
 ALT -19966220 07 SHA -12617149 07 ALP -17289476 03 DR -42811324 01 DP -15894875-04 ASO -96773534-01
 HGE -32014240 03 SVL -59718631 01 HNC -14019883 03 SIA -17393110 03

AREOCENTRIC EQUATORIAL COORDINATES
 X -28159035 06 Y -18703261 07 Z -65006682 06 DX -10731629 01 DY -40344328 01 DZ -10989137 01
 R -19999999 07 DEC -18967598 02 RA -98561966 02 V -43169362 01 PTH -82615601 02 AZ -30389252 03
 R -20000000 07 LAT -18967598 02 LON -21477832 03 VP -13459592 03 PTP -18227299 01 AZP -27013177 03
 XAS -99052002 02 DEE -24866261 02 RAS -14029968 03 OES -15866608 02 LOE -21524835 03 LOS -25651603 03

AREOCENTRIC CONIC
 EPOCH OF PERICENTER PASSAGE 235723202427202403527522 J.D.= 2438958.55775496 JULY 17, 1965 01 23 10.029
 SMA -23114860 04 ECC -11133878 03 B -25734764 06 SLR -28651617 08 APO -00000000 00 RCA -25504654 06
 VA -43119555 01 C3 -18992960 02 C1 -11096716 07 TFP -45904197 06 TF -23042723 03 LTF -23039799 03
 TA -83125959 02 HTA -90914614 03 EA -11502370 03 MA -49063444 05 FFI -22511424 03
 ZAE -16848501 03 ZAP -14587674 03 ZAC -98962530 02 DEF -10293611 01 IR -51987771 04 GP -86749242 01

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				ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE			
X .19755248 07	Y .29516781 06	Z -.10088414 06	DX -.41443964 01	DY -.11774092 01	DZ -.27133758 00		
INC .17957811 03	LAN .34183582 03	APF .56319054 02	HX .15201859 00	HY -.98332328 00	MZ .99828441-01		
MX -.34867790-01	MY -.10627491 00	MZ -.99372528 00	PX .26914820 00	PY -.95859092 00	PZ .93073576-01		
QX -.96246739 00	QY -.26421408 00	QZ .62027644-01	RX -.60465938-01	RY -.17183313-01	RZ -.90832227 00		
BX .27778184 00	BY -.95617918 00	BZ .92512715-01	TX -.27335371 00	TY .96191359 00	TZ .00000000 00		
SXI -.96001118 00	SYI -.27281309 00	SZI .62861092-01	DAI .36040509 01	RAI .19586392 03			
SXD -.96484594 00	SYD .25559374 00	SZD .61189193-01	DAO .35080731 01	RAO .19483720 03			
ETE .32002128 02	ETS .14294957 03	ETC .23281654 03					
				T VECTOR IN EARTH EQUATOR PLANE			
BTO -.25623962 06	BRD -.23855108 05	B .25734764 06	THA .18531872 03				
				ALL VECTORS REFERENCED TO ECLIPTIC PLANE			
X .19755248 07	Y .23066303 06	Z -.20999099 06	DX -.41443964 01	DY -.97225593 00	DZ -.71730604 00		
INC .15038975 03	LAN .35595333 03	APF .70857026 02	HX .15201845 00	HY -.86242015 00	MZ .48281090 00		
MX -.34867758-01	MY -.49286372 00	MZ -.86940663 00	PX .26914808 00	PY -.84242503 00	PZ .46677369 00		
QX -.96246739 00	QY -.21772405 00	QZ .16202702 00	RX .16181709 00	RY -.37972899-01	RZ -.98508989 00		
BX .27778195 00	BY -.84043626 00	BZ .46530000 00	TX -.22845947 00	TY .97355343 00	TZ .00000000 00		
SXI -.96001119 00	SYI .22528158 00	SZI .16621285 00	DAI .95676999 01	RAI .19320638 03			
SXD -.96484594 00	SYD .21014895 00	SZD .15782011 00	DAO .90808550 01	RAO .19228744 03			
ETE .94838743 02	ETS .16578618 03	ETC .25565315 03					
				T VECTOR IN ECLIPTIC PLANE			
BTC -.22689608 06	BRC -.12143301 06	B .25734764 06	THA .20815534 03				
				ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET			
X -.28159081 05	Y .19730736 07	Z -.16639353 06	DX .10731639 01	DY .41327579 01	DZ .63605370 00		
INC .14925252 03	LAN .90054571 02	APF .73760504 02	HX .84781782 00	MY .16353827 00	MZ .50443871 00		
MX .51125324 00	MY .48695213-02	MZ .85942896 00	PX .82487204 00	PY .28043826 00	PZ .49085507 00		
QX .24125624 00	QY .95987187 00	QZ .14297356 00	RX .37050400-01	RY .14266319 00	RZ .98908047 00		
BX .82267265 00	BY .28904839 00	BZ .48955159 00	TX .96788319 00	TY .25140034 00	TZ .00000000 00		
SXI .24865517 00	SYI .95731436 00	SZI .14737645 00	DAI .84749191 01	RAI .28456039 03			
SXD .23383784 00	SYD .96235193 00	SZD .13855914 00	DAO .79644786 01	RAO .28365738 03			
ETE .56350271 02	ETS .16729771 03	ETC .25716468 03					
				T VECTOR IN ORBIT PLANE OF TARGET			
BTO -.22361396 05	BRD -.12737582 06	B .25734764 06	THA .20966686 03				
				ALL VECTORS REFERENCED TO AREOCENTRIC EQUATOR PLANE			
X -.28159081 05	Y .18703260 07	Z .65006682 06	DX .10731639 01	DY .40344325 01	DZ .10989137 01		
INC .14172193 03	LAN .12438140 03	APF .11477325 03	HX .84781858 00	MY .55648232-01	MZ .52735849 00		
MX .51125371 00	MY .34981902 00	MZ .78501359 00	PX .82487278 00	PY .56678651-01	PZ .56246990 00		
QX .24125622 00	QY .93510115 00	QZ .25957899 00	RX .65441985-01	RY .24595948 00	RZ .96708837 00		
BX .82267267 00	BY .65075067 01	BZ .56477866 00	TX .96437878 00	TY .25712242 00	TZ .00000000 00		
SXI .24865516 00	SYI .93455437 00	SZI .25451664 00	DAI .14744945 02	RAI .28489939 03			
SXD .23383781 00	SYD .93557250 00	SZD .26462050 00	DAO .15344397 02	RAO .28403305 03			
ETE .62416562 02	ETS .17336400 03	ETC .26323097 03					
				T VECTOR IN TRUE TARGET EQU. PLANE			
BTT -.20890083 05	BRT -.15029387 05	B .25734764 06	THA .21973316 03				

230 DAYS 10 HRS. 21 MIN. 2.327 SEC. 235723542666202651147325 J.D.= 2438958.56179771 JULY 17,1965 01 28 59.322

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GEOCENTRIC			EQUATORIAL COORDINATES					
X -.21527995 09	Y -.33370439 08	Z -.14273386 08	DX -.11678219 02	DY -.24713026 02	DZ -.10779755 02			
R .21831805 09	DEC .37486067 01	RA .18881126 03	V .29382266 02	PTH .32988816 02	AZ .11331624 03			
AL .21831805 09	LAT .37486067 01	LON .23189716 03	VE .15863321 05	PIE .57782545-01	AZE .26996476 03			
XS -.62405360 08	YS .12716259 09	ZS .55144967 08	OX .26664940 02	DYS .11125287 02	DZS .48248740 01			
XH .35175000 06	YH .16575950 06	ZH .11054600 06	DXH .45601406 00	DYH .79613352 00	DZH .32725155 00			
XT -.21534786 09	YT .33127602 08	ZT .12498221 08	DXT .74922993 01	DYT .23570404 02	DZT .11052582 02			
RS .15203851 09	VS .29292839 02	RH .40425820 06	VH .97316541 00	RT .21834967 09	VT .17809815 02			
GED -.37740790 01	ALT .21831167 09	LDS .15925455 03	RAS .11616865 03	RAN .33476820 03	LDM .21054099 02			
DOT .35000000 02	DT .95999999 03	DR .15997919 02	SHA .21117487 09	DES .21266347 02	DEM .15869899 02			
GCL .10175299 03	HCL .18001457 03	ICL .32801724 03						
HELIOCENTRIC			ECLIPTIC COORDINATES					
X -.15279459 09	Y -.17489919 09	Z .18206950 06	DX .14986720 02	DY -.14835227 02	DZ -.57264804-01			
R .23224114 09	LAT .44918739-01	LON .22885904 03	V .21087651 02	PTH .35678491 01	AZ .90158692 02			
AL .62405360 08	YS .13806800 09	ZS .36950000 03	DXS .26664940 02	DYS .12126476 02	DZS .25510788 03			
XT -.15286250 09	YT .17468628 09	ZT .62649500 05	DXT .19172641 02	DYT .13895480 02	DZT .76217288 00			
LTE .13924623-03	LDE .29426664 03	LTT .15469524-01	LDT .22881184 03	RSI .23212549 09	VSI .23690830 02			
EPS .39289430 02	ESP .65407611 02	SEP .75302954 02	EPH .66719594-01	EMP .38905585 02	MEP .14102778 03			
MPS .39355420 02	MSP .65497911 02	SMP .75146666 02	SEM .14310437 03	EMS .36804360 02	ESM .91192092-01			
EPF .97133889 02	ETP .82800138 02	TEP .65610602-01	TPS .62814820 02	TSP .55071707-01	STP .11712955 03			
SEI .75244898 02	STE .39300298 02	EST .65454801 02	RPH .21863250 09	RPT .25337335 06	SPH .39287556 02			
SAC .89948905-10								
GCE .25824700 03	GCT .46264244 02	SIP .62050924 02	CPT .47739595 02	SIN .46975699 02				
REP .21831805 09	VFP .29382266 02	CPE .91154633 02	CPS .82009783 02					
AREOCENTRIC			ECLIPTIC COORDINATES					
X .67995585 05	Y -.21290933 05	Z .11940061 05	DX .44185920 01	DY .93974701 00	DZ .70490804 00			
R .25337335 06	DEC .28115064 02	RA .28768893 03	V .43476369 01	PTH .00000000 00	AZ .28059274 03			
AL .24999535 06	SHA .22549665 06	ALP .16413254 03	DR .46250916-07	DP .98313903-03	ASD .76389518 00			
HGE .32071056 03	SVL .28109262 02	HNG .58804424 02	SIA .96369993 02					
SAC .89948905-10								
			AREOCENTRIC EQUATORIAL COORDINATES					
X .20844761 06	Y .13603512 05	Z .14339664 06	DX .10420695 01	DY .40671214 01	DZ .11289647 01			
R .25337335 06	DEC .34468204 02	RA .37338876 01	V .43476368 01	PTH .29456472-06	AZ .25164103 03			
R .25337335 06	LAT .34468205 02	LON .54251868 02	VP .18982457 02	PJP .44976981-07	AZP .26586315 03			
RAE .10224721 03	DFE .24671811 02	RAS .14291916 03	DES .15017378 02	LOE .15276519 03	LOS .19343714 03			
			AREOCENTRIC CONIC					
EPDCH OF PERICENTER PASSAGE			235723542666202651074600 J.D.= 2438958.56179770 JULY 17,1965 01 28 59.321					
SMA -.23152535 04	ECC .11043655 03	Ø .25567811 06	SLR .28235053 08	APD .00000000 00	ACA .25337335 06			
YN .43084458 01	EQ .18542705 02	CI .11015753 07	TFP .65103445-02	TF .23043127 03	LTF .23040201 03			
TA .75613208 05	HTA .90518817 02	EA .00000000 00	FA .90518817 02	TA .23043127 03	TFA .23043127 03			
ZAE .16978730 03	ZAP .14321075 03	ZAC .98950581 02	DEF .10376917 01	IR .52012245 04	GP .84832144 01			

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ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
X -67905585 05 Y -24283757 06 Z +24835528 05
INC -17531446 03 LAN +34281734 03 APF +57372230 02
WX -34382493-01 WY -11191702 00 WZ -95920403 00
QX -96280351 00 QY -26281449 00 QZ +62753035-01
BX +27671320 00 BY -95599888 00 BZ -97447246-01
SXI -96033725 00 SYI -27140216 00 SZI +63638020-01
SKD -96519082 00 SYD -25412526 00 SZD +61862898-01
ETE +63880492 02 ETS +14413840 03 ETC +23280735 03
BTQ --25445631 06 BRQ --24965733 05 B --25567811 06
THA +18560358 03 T VECTOR IN EARTH EQUATOR PLANE
ALL VECTORS REFERENCED TO ECLIPTIC PLANE
X +67905585 05 Y -21290933 06 Z +11940061 06
INC -15010870 03 LAN +35604388 03 APF +71013729 02
WX -34382493-01 WY -49715855 00 WZ -86597247 00
QX -96280351 00 QY -21615121 00 QZ +16213590 00
BX +27671319 00 BY -83830711 00 BZ +46975632 00
SXI -96033725 00 SYI -22375123 00 SZI +16639635 00
SKD -96519082 00 SYD -20893347 00 SZD +15786215 00
ETE +69235305 02 ETS +16698522 03 ETC +25565417 03
BTC --22479983 06 BRC --12180449 06 B +25567811 06
THA -20845039 03 T VECTOR IN ECLIPTIC PLANE
ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET
X +20844761 06 Y +70722138 05 Z +12548319 06
INC -14896956 03 LAN +90114076 02 APF +73889917 02
WX +51549237 00 WY +10263728-02 WZ -85689357 00
QX +23968836 00 QY -96025502 00 QZ +14304100 00
BX +82048545 00 BY +78780589 00 BZ +49393461 00
SXI +24712597 00 SYI +95768820 00 SZI +14751961 00
SKD +23222710 00 SYD +96274309 00 SZD +13855065 00
ETE +70745116 02 ETS +16849503 03 ETC +25716398 03
BTD --22151246 06 BRD --12768525 06 B +25567811 06
THA -20996021 03 T VECTOR IN ORBIT PLANE OF TARGET
ALL VECTORS REFERENCED TO AREOCENTRIC EQUATOR PLANE
X -20844761 06 Y -13609502 05 Z +14339664 06
INC -14148794 03 LAN +12412029 03 APF -11464696 03
WX -51549237 00 WY +34928090 00 WZ -78247774 00
QX +23968836 00 QY +45484863 00 QZ -14304100 00
BX +82048545 00 BY +62158088-01 BZ +56827810 00
SXI +24712597 00 SYI -93495411 00 SZI -25453787 00
SKD +23222710 00 SYD -93592642 00 SZD -26478719 00
ETE +76774132 02 ETS +17452404 03 ETC +26319299 03
BTM --20687620 06 BTR --15024492 06 B +25567811 06
THA -21598922 03 T VECTOR IN TRUE TARGET EQU. PLANE
230 DAYS 10 HRS. 28 MIN. 31.060 SEC. 235723542666202651705722 J.D.= 2438958.56699143 JULY 17.1965 01 36 28.060

CASE 1

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GEOCENTRIC

EQUATORIAL COORDINATES

X -21528519 09 Y -33381530 08 Z -14278223 08
R -21832523 09 DEC -37497550 01 RA +18881394 03
XS -62497325 08 YS +12715759 09 ZS +55142802 08
XM +35195200 06 YM -16540250 06 ZM -11039887 06
XMT -21535122 09 YMT -33138180 08 ZMT -14303181 08
GSD -37752351 01 ALT +21031086 09 LOS +15738492 03
DIUT +35000000 02 DI +95999999 03 DR +15997706 02
GCL -10175268 03 HCL +18001457 03 TCL +32779482 03
HELIOCENTRIC EQUATORIAL COORDINATES
X -15278787 09 Y -17490584 09 Z +1820450 06
R -23224173 09 LAT +44971615-01 LON +22886137 03
XT +15285389 09 YT +17469252 09 ZT +62328000 05
LTE +13924627-03 LDE +79427160 03 LDT +15184845-01
EPI +96693188 02 ELP +83240775 02 TEP +66351990-01
SET -75242820 02 STE +39300033 02 EST +65457144 02
GCE +25824731 03 GCT +46042143 02 SGP +61654202 02
REP -21832523 09 VEP +29382748 02 CPE +91153933 02
AREOCENTRIC ECLIPTIC COORDINATES
X -66027186 05 Y -21333097 06 Z +11971690 06
R +25338079 06 DEC +34328284 02 RA +28719760 03
ALT +25000279 06 SHA +22469740 06 ALP -16379697 03
HGE +32071114 03 SVL -28189059 02 HNG +58310008 02
SAC +89948448-10
EPOCH OF PERICENTER PASSAGE 235723542666202651705722 J.D.= 2438958.56197777 JULY 17.1965 01 28 59.327
SMA -23152520 04 ECC +11043662 03 B +25567811 06 SLR +28235069 08
VH +63086471 01 C3 +18562716 02 C1 -11015757 07 TFP +44873269 08
TA +44115593 00 HFA +90518817 02 EA +43718589 00 MA +47844590 02
ZAE +16978623 03 ZAP +14320822 03 ZAC +98950570 02 DEF +10378332 01

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ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
X .66027186 05 Y -.24335024 06 Z .24957948 05 DX -41860009 01 DY -.11423338 01 DZ -.27279727 00
INC .17331641 03 LAN .34281740 03 APF -.57372299 02 MX -.96483854 00 MY -.25542721 00 MZ -.61996301-01
WX -.34382542-01 WY -.11119166 00 WZ -.99320404 00 PX -.26800593 00 PY -.95841796 00 PZ -.98019493-01
QX -.96280352 00 QY -.26281441 00 QZ -.62752809-01 RX -.61237934-01 RY -.17311628-01 RZ -.99797302 00
SX .27671310 00 SY -.95599890 00 SZ .97447249-01 TX -.27203348 00 TY .96228779 00 TZ -.00000000 00
SXI -.96033726 00 SYI -.27148208 00 SZI .63637859-01 DAI .36486460 01 RAI .19578530 03
SXO -.96519083 00 SYO -.25412519 00 SZO .61862732-01 DAD .35466737 01 RAD .19475064 03
EYE .46403853 02 ETS .14413949 03 ETC .23280735 03

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BTO -.25445630 06 BRO -.24965732 05 B .25567811 06 THA .18560358 03 T VECTOR IN EARTH EQUATOR PLANE
ALL VECTORS REFERENCED TO ECLIPTIC PLANE
X .66027186 05 Y -.21333097 06 Z .11971690 06 DX -41860009 01 DY -.93949461 00 DZ -.70476573 00
INC .15010870 03 LAN .35604390 03 APF -.71013749 02 MX -.96483853 00 MY -.20967486 00 MZ -.15850254 00
WX -.34382543-01 WY -.49716853 00 WZ -.86697250 00 PX -.26800593 00 PY -.84029881 00 PZ -.47124378 00
QX -.96280353 00 QY -.21615121 00 QZ -.16213572 00 RX -.16205567 00 RY -.37757728-01 RZ -.98605895 00
SX .27671310 00 SY -.03830713 00 SZ .46975694 00 TX .22691465 00 TY .97391465 00 TZ -.00000000 00
SXI -.96033727 00 SYI -.22375123 00 SZI .16649617 00 DAI .95783515 01 RAI .19311549 03
SXO -.96519084 00 SYO -.20853347 00 SZO .15786198 00 DAD .90828213 01 RAD .19219159 03
ETE .69250666 02 ETS .16698631 03 ETC .25565417 03

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BTC -.22479982 06 BRC -.12180449 06 B .25567811 06 THA .20845039 03 T VECTOR IN ECLIPTIC PLANE
ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET
X .20891516 06 Y .68848766 05 Z .12576220 06 DX .10418234 01 DY -.41749240 01 DZ .62174067 00
INC .14896956 03 LAN .90114108 02 APF -.73889938 02 MX .23334517 00 MY -.96237565 00 MZ .13922333 00
WX .51549223 00 WY .10266641-02 WZ -.85689364 00 PX .82268954 00 PY .27912237 00 PZ .49525006 00
QX .23968664 00 QY .96025498 00 QZ .14304079 00 RX .36859183-01 RY .14284037 00 RZ .98905915 00
DX .82048546 00 DY .28780600 00 DZ .49393454 00 TX .96828200 00 TY .24985992 00 TZ .00000000 00
SXI .24712624 00 SYI .95768817 00 SZI .14751939 00 DAI .84831998 01 RAI .28446922 03
SXO .23222739 00 SYO .96274305 00 SZO .13855045 00 DAD .79639766 01 RAD .28356149 03
ETE .70760472 02 ETS .16849611 03 ETC .25716397 03

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BTD -.22151248 06 BTD -.12768523 06 B .25567811 06 THA .20996020 03 T VECTOR IN ORBIT PLANE OF TARGET
ALL VECTORS REFERENCED TO AREOCENTRIC EQUATOR PLANE
X .20891516 06 Y .11778488 05 Z .14289000 06 DX .10418234 01 DY -.40671372 01 DZ .11291355 01
INC .14148793 03 LAN .12412032 03 APF .11464695 03 MX .23334517 00 MY .93586417 00 MZ .26402324 00
WX .51549222 00 WY .34928120 00 WZ -.78247710 00 PX .82268953 00 PY .53689705-01 PZ .56594993 00
QX .23968664 00 QY .93547851 00 QZ .25967335 00 RX .65045446-01 RY .24608678 00 RZ .96708273 00
DX .82048546 00 DY .62159233-01 DZ .56827807 00 TX .96679766 00 TY .2554313 00 TZ .00000000 00
SXI .24712624 00 SYI .93495399 00 SZI .25453804 00 DAI .14746214 02 RAI .28480576 03
SXO .23222738 00 SYO .93592631 00 SZO .26478736 00 DAD .15354317 02 RAD .28393513 03
ETE .76789494 02 ETS .17452513 03 ETC .26319299 03

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```

BTT -.20687620 06 BRT -.15024491 06 B .25567811 06 THA .21598922 03 T VECTOR IN TRUE TARGET EQU. PLANE
215542322304 214412017045 614626031734 601731460721 204537405516 601471557251 EARTH
21189808915 17481859072 000000000000 000000000000 INITIAL
220775513340 622733565354 217604377240 603414100141 601440622006 177435553167 MARS
235723543047 202007527522 000000000000 000000000000 END

```

U MATRIX FOR MAPPING FORWARD

ITERATION NUMBER 1

	X	Y	Z	DX	DY	DZ	MM	AU
X	-.14240628 06	.32833775 05	-.12743440 05	-.32201795-01	-.18749275-01	-.67970214-02	.00000000 00	.00000000 00
Y	-.55073909 05	.15447456 05	.55764004 04	-.13270882-01	-.73212989-02	-.26537464-02	.00000000 00	.00000000 00
Z	.82201993 05	-.21762468 05	-.91111089 04	.18877931-01	-.10989078-01	.43940406-02	.00000000 00	.00000000 00
DX	.37202730 08	-.51210248 07	-.23455938 07	.80190345 01	-.47951245 01	.16743353 01	.00000000 00	.00000000 00
DY	-.22529723 09	.54473221 08	.21533065 08	-.51212858 02	.29795080 02	-.11093539 02	.00000000 00	.00000000 00
DZ	.25683868 08	-.66539341 07	-.19942635 07	.59071261 01	-.33589278 01	.10381906 01	.00000000 00	.00000000 00
MM	.21474836 11	-.18468359 12	.74625055 11	-.14336800 06	.82944000 05	-.17536000 05	.00000000 01	.00000000 00
AU	-.46875000 00	-.71875000 00	-.29687500 00	.13411045-06	-.29429793-06	-.58207661-08	.00000000 00	-.10000000 01

COVARIANCE MATRIX AT IMPACT

ITERATION NUMBER 1

	X	Y	Z	DX	DY	DZ	MM	AU
X	.10637680 05	-.22420325 05	.12501384 05	-.29614010-02	-.15642850-02	-.28561890-02	-.21474894-07	-.11718800 02
Y	.22420325 05	.88347057 05	.47340614 05	-.28379472-01	.16219363-01	-.11518833-01	.18468361-06	-.17968771 02
Z	.12501384 05	.47340614 05	.34772597 05	-.11713876-01	.75178238-02	-.98098615-02	.74625093-07	-.74219149 01
DX	-.29614010-02	-.28379472-01	-.11713876-01	.20777808-07	-.12004298-07	.27655606-08	-.14335999-12	.33527526-05
DY	.15642850-02	.16219363-01	.75178238-02	-.12004298-07	.70263182-08	-.19162602-08	.82943994-13	-.73574456-05
DZ	-.28561890-02	-.11518833-01	-.98098615-02	.27655606-08	-.19162602-08	.29285971-08	-.17536011-13	-.14550667-06
MM	.21474894-07	.18468361-06	.74625093-07	-.14335999-12	.82943994-13	-.17536011-13	.99999998-18	.21195486-22
AU	-.11718800 02	-.17968771 02	-.74219149 01	.33527526-05	-.73574456-05	-.14550667-06	.21195486-22	.25000000 02

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ENCOUNTER PARAMETERS AND STATISTICS 65/07/17 012859.322

B .25568130 06 SMAA .32174637 03
 B.RD -.12768892 06 SMIA .83507688 02
 B.YD -.22151403 06 DEL T .37226349 02
 B.RC -.12180449 06 DEL B .33240677 03
 B.TC -.22479983 06 DEL S .16038767 03
 TL .23040207 03 DEL BR .10149567 03
 THETA .10699438 02 DEL BT .31653260 03

N MATRIX

	B.RD	B.TD	TL	C3	S.TS	S.RS
B.RD	.10301371 05	-.17613031 05	-.17042226 04	.14831570-01	.10897963-02	.74349552-03
B.TD	-.17613032 05	.10019289 06	.10369679 05	.95422841-01	-.69411049-02	-.30035637-02
TL	-.17042227 04	.10369679 05	.13858011 04	.51374437-02	-.62122480-03	-.24740746-03
C3	.14831570-01	.95422841-01	.51374439-02	.66149723-06	-.11622786-07	-.69983611-08
S.TS	.10897963-02	-.69411049-02	-.62122480-03	-.11622786-07	.53718767-09	.25609643-09
S.RS	.74349549-03	-.30035638-02	-.24740745-03	-.69983607-08	.25609644-09	.15636082-09

NORMALIZED N MATRIX

	B.RD	B.TD	TL	C3	S.TS	S.RS
B.RD	.99999999 00	-.54823672 00	-.45105381 00	.17967000 00	.46327063 00	.58582360 00
B.TD	-.54823674 00	.10000000 01	.88002785 00	.37065500 00	-.94612197 00	-.75884733 00
TL	-.45105381 00	.88002785 00	.10000000 01	.16968076 00	-.72000467 00	-.53149401 00
C3	.17967001 00	.37065500 00	.16968077 00	.99999998 00	-.01657117 00	-.68812661 00
S.TS	.46327065 00	-.94612197 00	-.72000467 00	-.61657117 00	.99999999 00	.88364239 00
S.RS	.58582357 00	-.75884738 00	-.53149399 00	-.68812657 00	.88364241 00	.99999999 00

DM/DDO MATRIX

	B.RD	B.TD	TL	C3	S.TS	S.RS
X	.24330516 05	.64995540 05	-.29677574 05	.13145920 00	-.11157073-01	.56606676-03
Y	.98666986 04	.27841637 05	-.11310281 05	.49878439-01	-.44800132-02	.25911632-03
Z	-.13503961 05	-.40737256 05	.16973777 05	-.74298253-01	.66583250-02	-.28822134-03
DX	-.59392066 07	-.13507410 08	.79610852 07	-.35645702 02	.26993601 01	-.10983936 00
DY	.38196933 08	.10563821 09	-.46808109 08	.20666396 03	-.17838378 02	.87565422 00
DZ	-.49480323 07	-.12244892 08	.53021018 07	-.23715798 02	.20346128 01	-.14765713 00
MU	.24638750 01	-.72774732 01	.14250950 00	.43780724-03	.10143746-04	-.60035930-05

DM/UQ MATRIX

	B.RD	B.TD	TL	C3	S.TS	S.RS
X	-.14297395 00	-.16182701 00	.22580654 00	.28314228-06	.43595943-07	-.25130324-07
Y	-.28136354-01	.98867121 00	.40707545-01	-.11281551-05	-.47139432-07	.27172928-07
Z	-.99584158 00	.32882216-02	-.35173765-01	.66306711-06	.55314040-07	.26337118-07
DX	-.63716973 04	-.16846420 04	.91028766 04	-.84913467 01	-.36826590-03	.21228334-03
DY	-.85946945 04	.40625272 04	.13368759 05	-.13992037 01	.15366673-02	-.88579189-03
DZ	.10419841 05	-.11039756 03	-.84511326 04	.12437810 01	.18654026 00	.26722055 00
MU	.24778832 01	-.87136382 01	.23160733-01	.43191793-03	.10281612-04	-.59267003-05

N MATRIX FOR MANEUVER CALCULATIONS

	B.RC	B.TC	TL
B.RC	.11291969 05	-.19957052 05	-.22881360-01
B.TC	-.19957053 05	.99202292 05	.11945784 00
TL	-.22881360-01	.11945784 00	.18564079-06

0000 140311 OFFLINE CONTROL I (27
 0000 14031 TERMINATE JOB
 0000 14031 END DATA - 10

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0000 141001 PAGE HEADING (23
0000 14100 IVENUS CHECK CASE - STEP MAP1
0000 14100 EPOCH (01
0000 14100 620900500,2332000
0000 14100 GEOCENTRIC POSITION AND VELOCITY AT EPOCH (02
0000 14100 X=-.14245297E7 Y=-.19398640E7 Z=-.10071723E6
0000 14100 DX=-.17446099E1 DY=-.24232877E1 DZ=-.11048412E0
0000 14100 OTHER OPTIONS AND CONSTANTS (03
0000 14100 ESTIMATE THESE PARAMETERS (04
0000 14100 X Y Z DX DY DZ MV AU
0000 14100 COVARIANCE MATRIX OF ESTIMATED PARAMETERS (10
0000 14100 DIAG=.1,.1,.1,.0001,.0001,.0001,-.1E-17,25.
0000 14100 MAP COVARIANCE MATRIX TO (24
0000 14100 621200700,0
0000 14100 OFFLINE CONTROL (27
0000 14100 KEY(3) KEY(5)
0000 14100 END DATA (10
    
```

```

INPUT COVARIANCE MATRIX OF ESTIMATED PARAMETERS ITERATION NUMBER 0
X Y Z DX DY DZ MV AU
X .99999999-01 -.00000000 00 .00000000 00 .00000000 00 .00000000 00 .00000000 00
Y .00000000 00 .99999999-01 .00000000 00 .00000000 00 .00000000 00 .00000000 00
Z .00000000 00 .00000000 00 .99999999-01 .00000000 00 .00000000 00 .00000000 00
DX .00000000 00 .00000000 00 .00000000 00 .99999999-04 .00000000 00 .00000000 00
DY .00000000 00 .00000000 00 .00000000 00 .00000000 00 .99999999-04 .00000000 00
DZ .00000000 00 .00000000 00 .00000000 00 .00000000 00 .00000000 00 .99999999-04
MV .00000000 00 .00000000 00 .00000000 00 .00000000 00 .00000000 00 .99999998-18
AU .00000000 00 .00000000 00 .00000000 00 .00000000 00 .00000000 00 .25000000 02
    
```

```

INPUT J MATRIX OF ESTIMATED PARAMETERS ITERATION NUMBER 0
X Y Z DX DY DZ MV AU
X .10000000 02 .00000000 00 .00000000 00 .00000000 00 .00000000 00 .00000000 00
Y .00000000 00 .10000000 02 .00000000 00 .00000000 00 .00000000 00 .00000000 00
Z .00000000 00 .00000000 00 .10000000 02 .00000000 00 .00000000 00 .00000000 00
DX .00000000 00 .00000000 00 .00000000 00 .10000000 05 .00000000 00 .00000000 00
DY .00000000 00 .00000000 00 .00000000 00 .00000000 00 .10000000 05 .00000000 00
DZ .00000000 00 .00000000 00 .00000000 00 .00000000 00 .00000000 00 .10000000 05
MV .00000000 00 .00000000 00 .00000000 00 .00000000 00 .00000000 00 .99999998 18
AU .00000000 00 .00000000 00 .00000000 00 .00000000 00 .00000000 00 .39999999-01
    
```

CASE 1 IBSYS-JPTRAJ-SPACE 022265 I

DOUBLE PRECISION EPHEMERIS TAPE - EPHEM1

```

GME .39880063 06 J .16234500-02 H -.57499999-05 D .78749999-05 RE .63781650 04 REM .63783112 04
G .65709998-19 A .88781796 29 a .88800194 29 C .88836976 29 OME .41780741-02 AU .14959850 09
GMM .49026293 04 GMS .13271411 12 GMV .32476627 06 GMA .42977368 05 GMC .37918700 08 GMJ .12670935 09
EGH .39880320 06 GGM .49027779 04 JA .29200000-02 HA .00000000 00 DA .00000000 00 RA .34170000 04
RADIATION PRESSURE INPUT
ARA .38300000 01 GB .38300000 00 MAS .19822000 03 GB1 .00000000 00 GB2 .00000000 00 SC .10310000 09
    
```

```

INJECTION CONDITIONS VENUS 2355740064120200000000 J.D.= 2437912.51634260 SEPT. 5, 1962 00 23 32.000
GEOCENTRIC X0=-.14245297 07 Y0=-.19398640 07 Z0=-.10071723 06 DX0=-.17446099 01 DY0=-.24232877 01 DZ0=-.11048412 00
CARTESIAN I0 .14120000 04 GHA .34951873 03 GHD .34361929 03
DATE OF RUN 022265A 14103 EARTH IS THE CENTRAL BODY FOR INTEGRATION COWELL EQUATIONS OF MOTION
    
```

```

PROBE IS OUT OF EARTH'S SHADOW
0 DAYS 0 HRS. 0 MIN. 0.000 SEC. 2355740064120200000000 J.D.= 2437912.51634260 SEPT. 5, 1962 00 23 32.000
    
```

```

EQUATORIAL COORDINATES
X -.14245296 07 Y -.19398639 07 Z -.10071723 06 DX -.17446098 01 DY -.24232875 01 DZ -.11048412 00
R .24088380 07 DEC -.23963223 01 RA .23370867 03 Y .29880081 01 PTH .89393198 02 AZ .62816414 02
R .24088380 07 LAT -.23963223 01 LUN .24418974 03 VE .17549891 03 PTE .97549786 00 AZE .27000472 03
XS -.14343227 09 YS .42810504 08 ZS .18564279 08 DXS -.87218607 01 DYS -.25899198 02 DZS -.11230749 02
XW -.27632510 06 YW .27943325 06 ZW -.81923071 05 DWS .72016653 00 DVW -.58854660 00 DZW -.27195625 00
MW -.08139502 08 YW -.41412272 08 ZW -.22861356 08 DWT .21205738 02 DWT .90988722 01 DWT .55559090 01
RS .15083166 09 VS .29546049 02 RW .40143479 06 VW .97560880 00 RT .10003094 09 VT .23734804 02
GED -.24126335 01 ALT .24024598 07 LOS .17386238 03 RAS .16338111 03 RAM .22532043 03 LDM .23580169 03
DUT .35000000 02 DT .19200000 04 DR .29878406 01 SHA -.22750333 07 DES .70698621 01 OEM -.11775395 02
CCL .60702505 02 MCL .18217293 03 TCL .33869367 03
    
```

GEOCENTRIC CNVIC

```

EPOCH OF PERICENTER PASSAGE 235574612773202330000000 J.D.= 2437903.65534361 AUG. 27, 1962 03 43 41.688
SHA -.46363768 05 EGC .11464198 01 B .25992350 05 SLR .14571132 05 APD .00000000 00 RCA .67885751 04
YH .29321057 01 EC .85972439 01 C1 .76210645 05 IFP .76559031 06 TF -.88609980 01 LTF -.08860057 01
TA .15011236 03 NTA .15072477 03 EA .25993107 03 MA .27740861 04
    
```

```

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
X -.14245296 07 Y -.19398639 07 Z -.10071723 06 DX -.17446098 01 DY -.24232875 01 DZ -.11048412 00
INV -.27271914 02 LAN .23836288 03 APF .20465420 03 MX .70574093 00 MY .54195777 00 MZ .45648036 00
INC .43960422 02 LAN .21470263 03 APF .23385448 03 MX .70574097 00 MY .31562129 00 MZ .63441003 00
WX -.39026905 00 MY .57429661 00 WZ .71981946 00 PX .16105420 00 PY .81227206 00 PZ .58063743 00
BX -.71196439 00 BY .53327496 00 BZ .45686411 00 TX .81161130 00 TY .58419782 00 TZ .00000000 00
SX1 -.30283033 00 SY1 .87814137 00 SZ1 .37051435 00 DAI .21747342 02 RAI .10902694 03
SX0 .58379930 00 SY0 .81105764 00 SZ0 .36930149-01 DAD .21164212 01 RAD .23425365 03
    
```

```

ALL VECTORS REFERENCED TO ECLIPTIC PLANE
X -.14245296 07 Y -.18198130 07 Z .67933540 06 DX -.17446098 01 DY -.22672196 01 DZ .86269591 00
INC .43960422 02 LAN .21470263 03 APF .23385448 03 MX .70574097 00 MY .31562129 00 MZ .63441003 00
WX -.39026905 00 MY .57429661 00 WZ .71981946 00 PX .16105420 00 PY .81227206 00 PZ .58063743 00
BX -.71196439 00 BY .53327496 00 BZ .45686411 00 TX .81161130 00 TY .58419782 00 TZ .00000000 00
    
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SXI --30283033 00	SYI --65825499 00	SZI --68928400 00	DAI --43573456 02	RAI .11470480 03
SXD --58379930 00	SYD --75880365 00	SZD --28878253 00	DAO --16785081 02	KAO .23242646 03
BVC .25821250 05 BRC --29777056 04 B .25992357 05 THA .35342171 03 T VECTOR IN ECLIPTIC PLANE				
ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET				
X --21047606 07	Y .99300253 06	Z .62163449 06	DX --26151033 01	DY -12088622 01
INC .46529416 02	LAM .14007295 03	APF .23072292 03	MX --14000245 00	MY --72132412 00
WX .46573257 00	WY .55647604 00	WZ .68798206 00	PX .82730393 00	PY .20775199-02
QX --31406365 00	QY .83080683 00	QZ --45939320 00	RX --24082291 00	RY .11130159 00
BK .13064194 00	BY .72572865 00	BZ --67545940 00	BK .41963216 00	BY .90774047 00
SXI --56802335 00	SYI .40812293 00	SZI --71463701 00	DAI --45613458 02	RAI .35697122 02
SXD --87521258 00	SYD .40449869 00	SZD --26529930 00	DAO .15384736 02	KAO .15519494 03
BYD .25564948 05 BRD --46834718 04 B .25994920 05 THA .34962039 03 T VECTOR IN ORBIT PLANE OF TARGET				
0 DAYS 8 HRS. 28 MIN. 56.404 SEC. 235575417563202063552055 J.D. = 2437912.86977319 SEPT. 5, 1962 08 52 28.404				
0 DAYS 8 HRS. 28 MIN. 56.404 SEC. 235575417563202063552055 J.D. = 2437912.86977319 SEPT. 5, 1962 08 52 28.404				
CHANGE OF PHASE OCCURS AT THIS POINT SUN IS THE CENTRAL BODY FOR INTEGRATION COMELL EQUATIONS OF MOTION				
95 DAYS 14 HRS. 14 MIN. 19.059 SEC. 235605301517202607404611 J.D. = 2438008.10961873 DEC. 9, 1962 14 37 51.059				
HELIOCENTRIC		EQUATORIAL COORDINATES		
X --51022567 07	Y .99263157 08	Z .46385104 08	DX --36494257 02	DY --58201960 01
R .10968494 09	LAT .25017435 02	LOX .92942489 02	V .37104845 02	PTH --77073126 01
XE .33012548 08	YE .13172989 09	ZE .57121301 08	DXE .29507273 02	DYE .60187031 01
XT --57926326 07	YT .97856442 08	ZT .44437148 08	DXT --35091492 02	DYT --26941798 01
LTE .22812463 02	LOE .75930981 02	LTE .24385356 02	LOT .93387681 02	RST .10762945 09
EPS .12889057 03	ESP .15695707 02	SEP .35413715 02	EPH .78819662-01	EMP .11190031 02
HPS .12896821 03	HSP .15754195 02	SMP .35277586 02	SEM .15551557 03	ESH .24425661 02
EPT .13652545 03	ETP .41616420 02	TEP .18581345 01	TPS .34321386 02	TSP .75043170 00
SET .34151903 02	STE .12978488 03	EST .16063212 02	RPM .51564817 08	RPT .25000001 07
SAC .22900140-09	GCT .32905271 03	SIP .34179292 02	CPT .72719292 02	SIN .72577188 02
REP .51206375 08	VEP .14973609 02	CPE .80195375 02	CPS .10226691 03	

HELIOCENTRIC CONIC		EQUATORIAL COORDINATES		
EPOCH OF PERICENTRAL PASSAGE 235607605023202427404611 J.D. = 2438037.01349749 JAN. 7, 1963 12 19 26.184				
SMA .12722493 09	ECC .19144548 00	B .12487168 09	SLR .12254896 02	PTO .31581844 02
TA .26606638 02	C3 .-10431455 04	E1 .40330759 10	TFP .-24972951 07	TF .12449715 03
VT .52176583 02	MTA .18000000 03	EA --41914634 02	MA .-36123921 02	MA .-10206039 09
		PER .28646125 03		
		IFI .95593274 02		

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE				
X --51022567 07	Y .99263157 08	Z .46385104 08	DX --36494257 02	DY --58201960 01
INC .18528344 01	LAM .33250316 03	APF .17232764 03	MX .99880594 00	MY --46875705-01
WX --14928005-01	WY .-42393483 00	WZ .90556966 00	PX .-81748708 00	PY .52667570 00
QX --57575346 00	QY .73681204 00	QZ --35442347 00	RX .-19593895 00	RY .12623598 00
BK .57575345 00	BY .73681203 00	BZ .35442346 00	TX .54159204 00	TY .84064094 00
DAP .13478637 02	RAP .14720786 03			
BTQ .11628279 09 BRQ --45510967 08 B .12487168 09 THA .33862552 03 T VECTOR IN EARTH EQUATOR PLANE				

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ALL VECTORS REFERENCED TO ECLIPTIC PLANE				
X --51022567 07	Y .10952328 09	Z .30665980 07	DX --36494257 02	DY --66610823 01
INC .18528344 01	LAM .33250316 03	APF .17232764 03	MX .99880594 00	MY --46875705-01
WX --14928005-01	WY .-42393483 00	WZ .90556966 00	PX .-81748708 00	PY .52667570 00
QX --57575346 00	QY .73681204 00	QZ --35442347 00	RX .-19593895 00	RY .12623598 00
BK .57575345 00	BY .73681203 00	BZ .35442346 00	TX .54159204 00	TY .84064094 00
DAP .13478637 02	RAP .14720786 03			
BTC .12480754 09 BRC --40013342 07 B .12487166 09 THA .35816372 03 T VECTOR IN ECLIPTIC PLANE				
ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET				
X .10522015 09	Y .30951192 08	Z .12362223 07	DX --15093336 02	DY .3785281 02
INC .42377940 01	LAM .20514100 03	APF .22340356 03	MX --28065725 00	MY .95702555 00
WX .31394514-01	WY .66895422-01	WZ .99726595 00	PX .36657532 00	PY .92400173 00
QX .92985851 00	QY .36397499 00	QZ --53687768-01	RX .-18637393-01	RY .47232232-01
BK .92985854 00	BY .36397900 00	BZ .53687770-01	TX .93020169 00	TY .-36704881 00
DAP .29105203 01	RAP .68466271 02			
BTD .12469110 09 BRD --67127396 07 B .12487166 09 THA .35691845 03 T VECTOR IN ORBIT PLANE OF TARGET				
95 DAYS 14 HRS. 14 MIN. 19.059 SEC. 235605301517202607404611 J.D. = 2438008.10961873 DEC. 9, 1962 14 37 51.059				
CHANGE OF PHASE OCCURS AT THIS POINT VENUS IS THE CENTRAL BODY FOR INTEGRATION COMELL EQUATIONS OF MOTION				
95 DAYS 14 HRS. 14 MIN. 19.059 SEC. 235605301517202607404611 J.D. = 2438008.10961873 DEC. 9, 1962 14 37 51.059				
GEOCENTRIC		EQUATORIAL COORDINATES		
X .38114804 08	Y .32486730 08	Z .10736197 08	DX .69869830 01	DY .11838899 02
R .51206375 08	DEC .12102736 02	RA .22082483 03	PTM .68707168 02	AZ .12448867 03
XE .51206373 08	LAT .12102736 02	LOX .28310597 03	VE .36465705 04	PTE .21920980 00
XS .33012548 08	YS .13172989 09	ZS .57121301 08	DXS .29507273 02	DYS .60187031 01
XH .21915150 06	YH .27889500 06	ZH .87959625 05	DXH .-85011053 00	DYH .95271705 00
XI .38805180 08	YI .33873445 09	ZI .12684154 08	DXH .-55842184 01	DYI .87128830 01
RS .14732764 09	VS .30227680 02	TH .36544041 06	VH .10777812 01	RT .53048469 08
GED .12182757 02	ALT .51199996 08	LDS .31861212 03	RAS .25593098 03	RAM .51842267 02
DIT .35000000 02	DT .36400000 04	DR .13951461 02	SMA .-29672879 08	DES .-22812463 02
CCL .27278050 03	MCL .17997891 03	TCL .-61833203 02		DEM .13927554 02
HELIOCENTRIC		EQUATORIAL COORDINATES		
X --51022567 07	Y .99263157 08	Z .46385104 08	DX --36494256 02	DY --58201960 01
R .10968494 09	LAT .25017435 02	LOX .92942489 02	V .37104845 02	PTH --77073126 01
XE .33012548 08	YE .13172989 09	ZE .57121301 08	DXE .29507273 02	DYE .60187031 01
XT --57926326 07	YT .97856442 08	ZT .44437148 08	DXT --35091492 02	DYT --26941798 01
LTE .22812463 02	LOE .75930981 02	LTT .24385356 02	LOT .93387681 02	RST .10762945 09
EPS .12889057 03	ESP .15695704 02	SEP .35413715 02	EPH .78819662-01	EMP .11190073 02
HPS .12896821 03	HSP .15754195 02	SMP .35277587 02	SEM .15551562 03	ESH .24425606 02
EPT .13652545 03	ETP .41616422 02	TEP .18580010 01	TPS .34321389 02	TSP .75039911 00
SET .34151904 02	STE .12978488 03	EST .16063212 02	RPM .51564818 08	RPT .25000000 07
SAC .22900140-09	GCT .32905271 03	SIP .34179292 02	CPT .72719292 02	SIN .72577188 02
REP .51206374 08	VEP .14973609 02	CPE .80195375 02	CPS .10226691 03	

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APHRODICENTRIC EQUATORIAL COORDINATES
X .69037591 06 Y .14067148 07 Z .19479566 07 DX -.14027647 01 DY -.31260163 01 DZ -.43307926 01

APHRODICENTRIC CONIC
EPOCH OF PERICENTER PASSAGE 23560563423202757046611 J.D.= 2438013.29235961 DEC. 14, 1962 19 00 59.871

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
X .69037591 06 Y .14067148 07 Z .19479566 07 DX -.14027647 01 DY -.31260163 01 DZ -.43307926 01

ALL VECTORS REFERENCED TO ECLIPTIC PLANE
X .69037591 06 Y .20655587 07 Z .12275376 07 DX -.14027647 01 DY -.45909108 01 DZ -.27297031 01

ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET
X .21701731 07 Y -.11002964 06 Z .12362209 07 DX -.47923321 01 DY .11675437 00 DZ -.27414251 01

T VECTOR IN EARTH EQUATOR PLANE
B .57624974 05 THA .16008325 03
T VECTOR IN ECLIPTIC PLANE
B .57624970 05 THA .17088060 03

100 DAYS 19 HRS. 23 MIN. 26.074 SEC. 235605635514202411434156 J.D.= 2438013.32428326 DEC. 14, 1962 19 46 58.075

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GEUCENTRIC EQUATORIAL COORDINATES
X -.41051080 08 Y -.38291325 08 Z -.13671191 08 DX -.74626395 01 DY -.14910036 02 DZ -.72400016 0

HELIOCENTRIC EQUATORIAL COORDINATES
X -.21453306 08 Y .95594916 08 Z .44384672 08 DX -.37462473 02 DY -.11365090 02 DZ -.57040765 01

APHRODICENTRIC EQUATORIAL COORDINATES
X .30728787 05 Y .40000217 04 Z -.23859226 05 DX -.30269631 01 DY -.40196605 01 DZ -.45723879 01

APHRODICENTRIC CONIC
EPOCH OF PERICENTER PASSAGE 235605635514202411425970 J.D.= 2438013.32428326 DEC. 14, 1962 19 46 58.075

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
X .30728787 05 Y .40000217 04 Z -.23859226 05 DX -.30269631 01 DY -.40196605 01 DZ -.45723879 01

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BTQ -33373639 05 BRQ .35684405 05 B .48858741 05 THA .13308353 03 T VECTOR IN EARTH EQUATOR PLANE
ALL VECTORS REFERENCED TO ECLIPTIC PLANE
X .30728787 05 Y -.58220504 04 Z -.23481196 05 DX -.30269631 01 DY -.55069054 01 DZ -.25958396 01

100 DAYS 19 HRS. 26 MIN. 19.059 SEC. 235605635567202607404611 J.D.= 2438013.32628540 DEC. 14,1962 19 49 51.059

GEOCENTRIC EQUATORIAL COORDINATES
X -.41052372 08 Y -.38293904 08 Z -.13672441 08 DX -.74907017 01 DY -.14914270 02 DZ -.72176348 01

HELIOCENTRIC EQUATORIAL COORDINATES
X -.21549989 08 Y .95592949 08 Z .44383688 08 DX -.37490680 02 DY -.11370283 02 DZ -.56021254 01

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APHRODIDCENTRIC EQUATORIAL COORDINATES
X .30202689 05 Y .33043787 04 Z -.24648217 05 DX -.30555658 01 DY -.40230900 01 DZ -.45496177 01

EPOCH OF PERICENTER PASSAGE
SMA -.10964909 05 ECC .45667502 01 B .48858736 05 SLR .21771055 06 APD .00000000 00 RCA .39100092 05

BTQ -33373633 05 BRQ .35684404 05 B .48858736 05 THA .13308352 03 T VECTOR IN EARTH EQUATOR PLANE
ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
X .30202689 05 Y .33043787 04 Z -.24648217 05 DX -.30555658 01 DY -.40230900 01 DZ -.45496177 01

BTQ -39749463 05 BRQ .28410495 05 B .48858736 05 THA .14444512 03 T VECTOR IN ECLIPTIC PLANE
ALL VECTORS REFERENCED TO ECLIPTIC PLANE
X .30202689 05 Y -.67741585 04 Z -.23928315 05 DX -.30555658 01 DY -.55009941 01 DZ -.25735845 01

BTQ -41602410 05 BRQ .25620611 05 B .48858738 05 THA .14837339 03 T VECTOR IN ORBIT PLANE OF TARGET
ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET
X .59016583 03 Y -.32310610 05 Z -.22053972 05 DX -.60669888 01 DY .15147077 01 DZ -.26679470 01

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625535100735 625730503775 621602710213 601700276504 602465436600 575674117131 EARTH
 Z1172191649 17448304640 000000000000 INITIAL
 21727454543 214522350406 617601552025 602611220703 603400724620 603442741665 VENUS
 235605635567 202607404611 0000000000 END

U MATRIX FOR MAPPING FORWARD

	ITERATION						NUMBER			0		
	X	Y	Z	DX	OY	OZ	MV	AU				
X	.39773183 01	-.21087921 01	.10280934 01	.40090036-06	-.12381001-05	.60017031-06	.00000000 00	.00000000 00				
Y	-.42917053 00	.30646667 00	-.21520492 00	-.14571548-06	-.55214985-08	-.10965675-06	.00000000 00	.00000000 00				
Z	-.15201560 00	.16763731 00	-.13895442 00	.55276687-07	-.73585203-07	-.18714900-06	.00000000 00	.00000000 00				
DX	-.13987192 08	.68392637 07	-.33544449 07	-.16401137 01	-.40547289 01	.19595122 01	.00000000 00	.00000000 00				
OY	-.36523107 07	-.79537667 07	-.19422097 07	-.14248229 01	-.25076387 01	-.14931677 01	.00000000 00	.00000000 00				

DDPX*.14130

621200700.0000000

X* = -.36497317E+08 Y* = -.29912202E+08 Z* = -.94567338E+07 \$ 0 0 0 1
 DX* = -.73404306E+01 DY* = -.10826458E+02 DZ* = -.54166483E+01 \$ 0 0 0 2
 R01 243474261372210626001145/8 242576315205207014363212/8 \$ 0 0 0 3
 241566472073206622212017/8 214557475024161672110320/8 \$ 0 0 0 4
 215654616340152603525674/8 214635656735161265311433/8 \$ 0 0 0 5
 54303671621510350007741/8 17757373651714452000000/8 \$ 0 0 0 6
 R02 242576315205207014363212/8 242523635265207430612242/8 \$ 0 0 0 7
 241435426012206550201533/8 214453020170161507154504/8 \$ 0 0 0 8
 215474360771162436676672/8 214477343136161731064435/8 \$ 0 0 0 9
 544502533776511164214012/8 200507013020145560000000/8 \$ 0 0 0 10
 DDPX*.14130

R03 241566472073206622212011/8 241435426012206550201534/8 \$ 0 0 0 11
 240714243507205560121535/8 213443050761160413054706/8 \$ 0 0 0 12
 214501770732161554051315/8 212772213025157172636427/8 \$ 0 0 0 13
 544550652467511083591222/8 200406075474145460000000/8 \$ 0 0 0 14
 R04 214557475024161672110322/8 214453020170161507154504/8 \$ 0 0 0 15
 213443050767160413054706/8 166420613606133125331104/8 \$ 0 0 0 16
 167447242664134104655056/8 166434734635133525462314/8 \$ 0 0 0 17
 51755136273446464233410/8 550774612640515700000000/8 \$ 0 0 0 18
 R05 215654616340162603525674/8 215474360771162436676673/8 \$ 0 0 0 19
 214501770732161554051315/8 167447242664134104655056/8 \$ 0 0 0 20
 DDPX*.14130

170507327007135360406204/8 167667560104134725552126/8 \$ 0 0 0 21
 520724131133465015437024/8 154513072266121140000000/8 \$ 0 0 0 22
 R06 214695656735161265311433/8 214477343136161731064435/8 \$ 0 0 0 23
 212772213025157172636427/8 166434734635133525462314/8 \$ 0 0 0 24
 167667560104134725552126/8 166477114443133230565525/8 \$ 0 0 0 25
 521423551405466611542605/8 153566167242120740000000/8 \$ 0 0 0 26
 R07 54303671621510350007741/8 544502533776511164214012/8 \$ 0 0 0 27
 544550652467511083531222/8 51755136273446464233410/8 \$ 0 0 0 28
 520724131133465015437024/8 52142355140546611542605/8 \$ 0 0 0 29
 105447113564052351103530/8 000000000000000000000000/8 \$ 0 0 0 30
 DZ .18190670 07 .19553291 07 .48569464 07 .69472166 00 .15003603 01 .13430120 00 .00000000 00 .00000000 00
 MV -.94537556 09 -.23478037 10 -.26254072 10 -.12541165 04 -.32486109 04 -.38261904 04 .10000000 01 .00000000 00
 AU .14841206-01 .25948558-01 .20478152-01 .-2.3691501-08 .24669914-07 .16027430-07 .00000000 00 .10000000 01

CONDITIONS AFTER FORWARD MAPPING 62/09/05 002332.000 TD 62/12/07 000000.000

X* = -.36497317 08 Y* = -.29912202 08 Z* = -.94567338 07 DX* = -.73404306 01 DY* = -.10826458 02 DZ* = -.54166483 01

STANDARD DEVIATIONS

X = .14570403 06 Y = .10678034 06 Z = .62140492 05 DX = .22809497-01 DY = .49980155-01 DZ = .24672443-01

COVARIANCE MATRIX AFTER MAPPING

	ITERATION						NUMBER			0		
	X	Y	Z	DX	OY	OZ	MV	AU				
X	-.21229665 11	-.12831745 11	-.62849716 10	.29409537 04	-.68804511 04	-.33107333 04	-.94537555-09	.37103015 00				
Y	-.12831245 11	-.11402042 11	-.47906166 10	-.23922537 04	-.50635308 04	-.2555497 04	-.23478037-08	-.63871396 00				
Z	-.62849716 10	-.47906166 10	-.38614407 10	-.11643202 04	-.25758895 04	-.10125431 04	-.26254072-08	-.51195380 00				
DX	.29409537 04	.23922537 04	-.11643202 04	-.52027313-03	.11265490-02	.54346325-03	-.12541165-14	-.59228753-07				
OY	-.68804511 04	-.50635308 04	-.25758895 04	-.11265490-02	-.24980159-02	.11891137-02	-.32486109-14	.61674783-06				
OZ	-.33107333 04	-.2555497 04	-.10125431 04	.54346325-03	-.11891137-02	.68872945-03	-.38261903-14	-.40068574-06				
MV	-.94537555-09	-.23478037-08	-.26254072-08	-.12541165-14	-.32486109-14	-.38261903-14	.99999998-18	.00000000 00				
AU	.37103015 00	-.63871396 00	-.51195380 00	-.59228753-07	.61674783-06	.40068574-06	.00000000 00	-.25000000 02				

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DOUBLE PRECISION EPHEMERIS TAPE - EPHE41

GME .39860063 06 J .16234500-02 H -.57499999-05 D .78749999-05 RE .63781650 04 REM .63783112 04
 G .66719998-19 A .80781796 29 B .88800194 29 C .88836975 29 DME .41780741-02 AU .14959850 09
 'GMH .49026293 04 GMS .13271911 12 GNV .32476627 06 GMA .42977368 05 GMC .37918700 08 GMJ .12670935 09
 EGM .39860320 06 YGN .49027779 04 JA .29200000-02 HA .00000000 00 DA .00000000 00 RA .34170000 04
 RADIATION PRESSURE INPUT
 ARA .38300000 01 GB .38300000 00 HAS .19822000 03 GB1 .00000000 00 S02 .00000000 00 SC .10310000 09

INJECTION CONDITIONS VENUS 235605123440202000000000 J.D.= 2438009.50000000 DEC. 7.1962 00 00 00.000

GEOCENTRIC TO-.36497317 08 YO-.29912202 08 ZO-.94567338 07 DXO-.73404306 01 DYO-.10826458 02 DZO-.54166483 01
 CARTESIAN IO .00000000 00 OHA .75283983 02 GHO .75283983 02
 DATE OF RUN 022265A 1435 SUN IS THE CENTRAL BODY FOR INTEGRATION COWELL EQUATIONS OF MOTION

2 DAYS 14 HRS. 37 MIN. 51.431 SEC. 235605301517202667117775 J.D.= 2438008.10962304 DEC. 9.1962 14 37 51.431

HELIOCENTRIC EQUATORIAL COORDINATES

X -.51022667 07 Y .99263156 08 Z .46385103 08 DX -.36494255 02 DY -.58201981 01 DZ -.33262720 01
 R .10968849 09 LAT .25017435 02 LON .92942495 02 V .37104844 02 PTH .77073117 01 AZ .92101534 02
 XE .33012537 08 YE .13172989 09 ZE .57121303 08 OXE -.29507275 02 OYE .60187012 01 OZE .26093421 01
 XT -.57926456 07 YT .97856441 08 ZT .44437148 08 OXT -.35091491 02 OYT .26941837 01 OZT .10045197 01
 LTE .22812464 02 LOE .75930985 02 LIT .24385356 02 LOT .93387689 02 RST .10762945 09 VST .35209096 02
 EPS .12889057 03 ESP .15695705 02 SEP .35413716 02 EPM .79129368-01 EMP .11190247 02 MEP .16873040 03
 MPS .12896821 03 MSP .15754197 02 SMP .35277589 02 SEM .15951569 03 EMS .24425547 02 ESM .57674939-01
 EPT .13652552 03 ETP .41616357 02 TEP .18581082 01 TPS .34321416 02 TSP .75039911 00 STP .14492818 03
 SET .34151907 02 STE .12978487 03 EST .16063214 02 RPM .51564819 08 RPT .25000001 07 SPH .12888343 03
 S4C .22900140-09
 GCE .87219449 02 GCT .32905258 03 SIP .34179322 02 CPT .72719294 02 SIN .72577199 02
 REP .51206376 08 VEP .14973609 02 CPE .80195375 02 CPS .10226691 03

HELIOCENTRIC CONIC

EPOCH OF PERICENTER PASSAGE 235607605023202607117775 J.D.= 2438037.01350758 JAN. 7.1963 12 19 27.056
 SMA .12722491 09 FCC .19144538 00 B .12487167 09 SLR .12256195 09 APD .15158153 09 RCA .10288829 09
 VHI .26606642 02 C3 .10431457 04 C1 .40330757 10 TFP .24972956 07 TF .31513507 02 PER .28646119 03
 TA .52176596 02 MTA .18000000 03 EA .43934648 02 MA .36323937 02 TFI .26096230 01

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
 X -.51022667 07 Y .99263156 08 Z .46385103 08 DX -.36494255 02 DY -.58201981 01 DZ -.33262720 01
 INC .25099881 02 LAN .35794326 03 APF .14656952 03 MX .-99880594 00 MY .-35811855-01 MZ .-33229988-01
 NX .-14928002-01 NY .-42393484 00 WZ .90556967 00 PX .-81748733 00 PY .52667545 00 PZ .23002869 00
 OX .-57575317 00 OY .-73681227 00 OZ .-35442357 00 RX .-19593890 00 RY .12623585 00 RZ .-97245697 00
 BX .-57575314 00 BY .73681223 00 BZ .35442355 00 TX .54159255 00 TY .84064113 00 TZ .00000000 00
 DAP .13478631 02 RAP .14720788 03

BITQ .11628278 09 BRO .-45510973 08 B .12487167 09 THA .33862552 03 T VECTOR IN EARTH EQUATOR PLANE

ALL VECTORS REFERENCED TO ECLIPTIC PLANE
 X -.51022667 07 Y .10952327 09 Z .30665980 07 DX .-36494255 02 DY .-66630846 01 DZ .-73627058 00
 INC .18528344 01 LAN .33250317 03 APF .17232766 03 MX .-99880594 00 MY .-46075798-01 MZ .16240136-01
 NX .-14928002-01 NY .-28680308-01 WZ .99947717 00 PX .-81748730 00 PY .57593041 00 PZ .43166661-02
 OX .-57575309 00 OY .-81695946 00 OZ .-32043278-01 RX .-35288931-02 RY .24861228-02 RZ .-99999066 00
 BX .-57575310 00 BY .81695947 00 BZ .32043278-01 TX .57593579 00 TY .81749494 00 TZ .00000000 00

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DAP .24732883 00 RAP .144483480 03

BITC .12480753 09 URC .-40013344 07 B .12487165 09 THA .35816372 03 T VECTOR IN ECLIPTIC PLANE

ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET
 X .10522014 09 Y .30951206 08 Z .12362214 07 DX .-69869809 01 DY .-11838899 02 DZ .-59356140 01
 INC .62377892 01 LAN .20514101 03 APF .22340358 03 MX .28065739 00 MY .95702552 00 MZ .-73031394-01
 NX .-31394514-01 NY .66895417-01 WZ .99726596 00 PX .36657501 00 PY .92900175 00 PZ .50776340-01
 OX .-92985853 00 OY .36397868 00 OZ .-53687747-01 RX .-18637381-01 RY .-47232243-01 RZ .-99870991 00
 BX .92985865 00 BY .-36397873 00 BZ .53687754-01 TX .93020180 00 TY .-36704854 00 TZ .00000000 00
 DAP .-29105203 01 RAP .68466288 02

BITQ .12469108 09 BRO .-67127359 07 B .12487163 09 THA .35691845 03 T VECTOR IN ORBIT PLANE OF TARGET

2 DAYS 14 HRS. 37 MIN. 51.431 SEC. 235605301517202667117775 J.D.= 2438008.10962304 DEC. 9.1962 14 37 51.431
 CHANGE OF PHASE OCCURS AT THIS POINT VENUS IS THE CENTRAL BODY FOR INTEGRATION COWELL EQUATIONS OF MOTION
 PROBE IS LEAVING VENUS* SHADOW

2 DAYS 14 HRS. 37 MIN. 51.431 SEC. 235605301517202667117775 J.D.= 2438008.10962304 DEC. 9.1962 14 37 51.431

GEOCENTRIC EQUATORIAL COORDINATES

X .-38114804 08 Y .-32466733 08 Z .-10736199 08 DX .-69869809 01 DY .-11838899 02 DZ .-59356140 01
 R .51206376 08 DEC .-12102737 02 RA .22042483 03 V .14973609 02 PTH .68707164 02 AZ .12448867 03
 R .51206376 08 LAT .-12102737 02 LON .20310441 03 VE .36465785 04 PTE .21920980 00 ALE .26995161 03
 XS .-33012537 08 YS .-13172989 09 ZS .57121302 08 OXS .-29507274 02 OXS .-60187012 01 OXS .26093421 01
 XM .21915100 06 YM .27889500 06 ZM .87959750 05 OXM .-85011119 00 OYM .59271622 00 OYM .29599127 00
 YT .-38805183 08 YT .-33873448 08 ZT .-12684155 08 OYT .-55842176 01 OYT .-87128849 01 OYT .-16048224 01
 RS .14732764 09 VS .30227679 02 RM .36544016 06 VM .10777812 01 RT .53048472 08 VT .10472252 02
 GEO .-12182759 02 ALT .91199998 08 LOS .31861057 03 RAS .25593098 03 RAM .51840329 02 LOH .11451991 03
 DUT .35000000 02 DT .38400000 04 DR .13951460 02 SHA .-29672880 08 DES .-22812463 02 DEM .13927595 02
 CCL .27278050 03 HCL .17997891 03 TCL .61833078 02

HELIOCENTRIC EQUATORIAL COORDINATES

X -.51022667 07 Y .99263155 08 Z .46385103 08 DX -.36494254 02 DY -.58201981 01 DZ -.33262720 01
 R .10968849 09 LAT .25017435 02 LON .92942495 02 V .37104843 02 PTH .77073126 01 AZ .92101534 02
 XE .33012537 08 YE .13172989 09 ZE .57121302 08 OXE .-29507274 02 OXE .60187012 01 OXE .26093421 01
 XT -.57926456 07 YT .97856441 08 ZT .44437148 08 OXT -.35091491 02 OYT .26941837 01 OZT .10045197 01
 LTE .22812463 02 LOE .75930985 02 LIT .24385356 02 LOT .93387689 02 RST .10762945 09 VST .35209096 02
 EPS .12889057 03 ESP .15695707 02 SEP .35413716 02 EPM .79129368-01 EMP .11190133 02 MEP .16873051 03
 MPS .12896821 03 MSP .15754197 02 SMP .35277589 02 SEM .15951569 03 EMS .24425541 02 ESM .58816955-01
 EPT .13652552 03 ETP .41616357 02 TEP .18581082 01 TPS .34321416 02 TSP .75039911 00 STP .14492818 03
 SET .34151907 02 STE .12978487 03 EST .16063214 02 RPM .51564819 08 RPT .25000001 07 SPH .12888343 03
 S4C .22900140-09
 GCE .87219449 02 GCT .32905258 03 SIP .34179322 02 CPT .72719294 02 SIN .72577199 02
 REP .51206376 08 VEP .14973609 02 CPE .80195375 02 CPS .10226691 03

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APHRODIOCENTRIC EQUATORIAL COORDINATES
X .69037884 06 Y -14067148 07 Z .19479564 07 DX -14027633 01 DY -31260145 01 DZ -43307917 01

APHRODIOCENTRIC CONIC
EPOCH OF PERICENTER PASSAGE 2354056342320207711775 J.D. = 2438013.27230682 DEC. 14, 1962 19 01 00.493

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
X .69037884 06 Y -14067148 07 Z .19479564 07 DX -14027633 01 DY -31260145 01 DZ -43307917 01

ALL VECTORS REFERENCED TO ECLIPTIC PLANE
X .69037884 06 Y -20655583 07 Z -12275367 07 DX -14027633 01 DY -45909088 01 DZ -27297030 01

ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET
X .21701734 07 Y -11003277 06 Z -12362202 07 DX -47923297 01 DY -11075377 01 DZ -27414250 01

7 DAYS 19 HRS. 46 MIN. 58.744 SEC.
235405635514202537267173 J.D. = 2438013.32429101 DEC. 14, 1962 19 46 58.745

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GEOCENTRIC EQUATORIAL COORDINATES
X -.41051080 08 Y -.38291332 08 Z -.13671194 08 DX -74625149 01 DY -14909948 02 DZ -72400025 01

HELIOCENTRIC EQUATORIAL COORDINATES
X -.21653526 08 Y .95949411 08 Z -44384671 08 DX -37462349 02 DY -11365006 02 DZ -57040790 01

APHRODIOCENTRIC EQUATORIAL COORDINATES
X .30731973 05 Y .40003297 04 Z -.23860705 05 DX .30268405 01 DY -40195693 01 DZ -45723873 01

APHRODIOCENTRIC CONIC
EPOCH OF PERICENTER PASSAGE 235605635514202537267173 J.D. = 2438013.32429101 DEC. 14, 1962 19 46 58.745

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
X .30731973 05 Y .40003297 04 Z -.23860705 05 DX .30268405 01 DY -40195693 01 DZ -45723873 01

ALL VECTORS REFERENCED TO ECLIPTIC PLANE
X .30731973 05 Y .20822719 03 Z -23860705 05 DX -44519376 01 DY -59120630 01 DZ -67251588 00

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BTQ -33376521 05 BRD .35686536 05 B .48862267 05 THA -13308429 03 T VECTOR IN EARTH EQUATOR PLANE

X .30731973 05 Y -.58223563 04 Z -.23482676 05 DX -.3028405 01 DY -.55068224 01 DZ -.25958754 01
INC -13464238 03 LAN .21712663 03 APF .23754634 03 MX -.44519377 00 MY -.80995448 00 MZ -.38180656 00
WX -.42945021 00 WY .56728648 00 WZ -.70267956 00 PX .78573216 00 PY -.14888166 00 PZ -.60038753 00
OX -.44519377 00 OY -.80995448 00 OZ -.38180656 00 RX -.15308931 00 RY .48018924 00 RZ -.86730243 00
BX -.86414488 00 BY .32097534-01 BZ -.50221853 00 TX -.95275257 00 TY .30374750 00 TZ .00000000 00
SX1 -.26234738 00 SY1 -.82289474 00 SZ1 -.50400203 00 DA1 -.30265128 02 RA1 .25231717 03
SX0 -.60643401 00 SY0 -.75770547 00 SZ0 -.24108124 00 DA0 -.13950363 02 RAD .23132775 03
ETE .31365265 03 ETS .40592005 02 ETC .26213010 03

BTC -.39752716 05 BRC .28412022 05 B .48862270 05 THA .14444588 03 T VECTOR IN ECLIPTIC PLANE

X .16000420 04 Y -.32573250 05 Z -.21591972 05 DX -.60658784 01 DY .14841501 01 DZ -.26884667 01
INC -13723007 03 LAN .13851527 03 APF -.23438643 03 MX -.89218155 00 MY -.21829178 00 MZ -.39524777 00
WX .44982056 00 WY .50870284 00 WZ -.73408642 00 PX .40908683-01 PY -.83280857 00 PZ -.55204743 00
OX -.89218155 00 OY .21829177 00 OZ -.39542477 00 RX .50638509 00 RY -.18010562-01 RZ -.86211930 00
BX .23526740 00 BY -.86039665 00 BZ -.45206955 00 TX .35444453-01 TY .99936869 00 TZ .00000000 00
SX1 -.86157454 00 SY1 .30643560-01 SZ1 -.50670528 00 DA1 -.30444619 02 RA1 .17796302 03
SX0 -.87948921 00 SY0 .39543584 00 SZ0 -.26495350 00 DA0 -.15364189 02 RAD .15579524 03
ETE .30926123 03 ETS .61084645 02 ETC .27339349 03

BTD -.41605749 05 BRD .25621909 05 B .48862269 05 THA .14837416 03 T VECTOR IN ORBIT PLANE OF TARGET

7 DAYS 19 HRS. 49 MIN. 51.431 SEC. 235605635567202667117775 J.D.= 2438013.32628971 DEC. 14,1967 19 49 51.431

GEOCENTRIC EQUATORIAL COORDINATES

X -.41052371 08 Y -.38293907 08 Z -.13672442 08 DX -.74905246 01 DY -.14916415 02 DZ -.72176794 01
R .57781104 08 DEC -.13687418 02 RA .22300892 03 Y .18183384 02 PTH .68464354 02 AZ .11972300 03
A .57781103 08 LAT -.13687418 02 LON .22300892 03 VE .4080464 04 PRE .23105819 00 AZE .26095601 03
XS -.19592373 08 YS -.13388685 09 ZS .58056130 08 OXS .29999978 02 DYS .35439852 01 DZS .15355085 01
XM -.22548700 06 YM .28642250 06 ZM .12792487 06 OXM .85825437 00 DYX .53245234 00 OZM .13055600 00
XI -.41082577 08 YI .38297212 08 ZI .13647794 08 OXI .44351348 01 OYI .10891182 02 DZI .26880179 01
RS .14726152 09 VS .30247584 02 RM .38632500 06 VM .10184061 01 RT .57798435 08 VT .12058465 02
GEO .13777152 02 ALT .57774727 08 LOS .24121233 03 RAS .28167468 03 RAM .12821110 03 LOM .10774934 03
DUI .35000000 02 DT .24000003 03 DR .16913991 02 SHA .35391056 04 BES .23221815 02 DEH .19337945 02
CCL .27338252 03 MCL .17980012 03 TCL .32423443 03

HELIOCENTRIC EQUATORIAL COORDINATES

X -.21459998 08 Y .95592947 08 Z .44383687 08 DX -.37490503 02 DY .11370190 02 DZ .56821710 01
R .10195675 09 LAT .24371643 02 LON .10265216 03 Y .3958491 02 PTH -.72123917 01 AZ .95825938 02
A .19592373 08 VE .13388685 09 ZE .58056130 08 OXE .29999978 02 DYE .35439852 01 DZE .15355085 01
XI .21490205 08 YI .95596442 08 ZI .44408336 08 OXI .34435113 02 OYI .17347198 01 OZI .11325095 01
LIE .23221815 02 LOE .81674692 02 LIT .24382856 02 LOD .10267042 03 RST .10757002 09 VST .35228410 02
EPS .12301850 03 ESP .19210845 02 SEP .37770653 02 EPM .37807118 00 EPN .80703716 02 MEP .98918229 02
MPS .12339618 03 MSP .19108037 02 SMP .37495777 02 SEM .13664018 03 EMS .43258800 02 ESM .10231507 00
EPT .11709311 03 ETP .62872362 02 TEP .34970568-01 TFS .10908964 03 TSP .18504685-01 SIP .70170748 02
SET .37797815 02 STE .12297542 03 EST .19226759 02 RPM .57846254 03 RPI .39126494 05 SPN .12301217 03
SAC .23815338-09
GCL .86617479 02 GCT .23085190 03 CPT .12031658 03 SIN .11119915 03
REP .57781104 08 VEP .18183384 02 CPE .80314978 02 CPS .10271903 03

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APHRODIOCENTRIC EQUATORIAL COORDINATES

X .30206808 05 Y .33059017 04 Z -.24648340 05 DX -.30553899 01 DY -.40229929 01 DZ -.45496615 01
R .39126984 05 DEC -.39049491 02 RA .62455221 01 Y .64987717 01 PTH .14101310 01 AZ .21265003 03
ALT .32924984 05 SHA .26807055 05 ALP .18171916 02 DR .16736299 00 UP .99523646-02 ASD .91174290 01
HGE .23698150 03 SVL .33359017 02 HNG .11393792 03 SIA .10797568 03
SAC .23815338-09

APHRODIOCENTRIC CONIC

EPOCH OF PERICENTER PASSAGE 2356056355614207537325475 J.D.= 2438013.32429103 DEC. 14,1967 19 46 58.745

SHA -.10964917 05 ECC .45670611 01 B .49862266 05 SLR .2176185 06 APO .00000000 00 RCA .39112531 05
YH .54423037 01 C3 .29618670 02 C1 .26592331 06 E1 .17268546 03 TF .78242908 01 LTF .77888724 01
VA .17194770 01 PTA .10264791 03 EA .13765490 01 MA .49108403 01 NI .78262896 01 FFI .78262896 01
ZAE .13776067 03 ZAP .39973174 02 ZAC .71965343 02 DEF .25295826 02 IR .13206248 05 GP .30444612 02

X .30206808 05 Y .33059017 04 Z -.24648340 05 DX -.30553899 01 DY -.40229929 01 DZ -.45496615 01
INC .11477125 03 LAN .20822719 03 APF .22221180 03 MX -.46857002 00 MY -.59400908 00 MZ .63390773 00
WX .42945024 00 WY .80000898 00 WZ -.41899656 00 PX .78573216 00 PY .10227740 00 PZ .61005274 00
OX -.44519381 00 OY .59120634 00 OZ .67251579 00 RX .33778181 00 RY .71389434 00 RZ .81339893 00
BX .86414484 00 BY .22924561 00 BZ .44719571 00 TX .90492326 00 TY .42763471 00 TZ .00000000 00
SX1 .26234738 00 SY1 .82289480 00 SZ1 .50400192 00 DA1 .30265120 02 RA1 .25231716 03
SX0 .60643400 00 SY0 .75770550 00 SZ0 .24108117 00 DA0 -.13950360 02 RAD .23132775 03
ETE .30228987 03 ETS .29228412 02 ETC .25076851 03

BTQ -.33376518 05 BRQ .35686538 05 B .48862266 05 THA .13308428 03 T VECTOR IN EARTH EQUATOR PLANE

X .30206808 05 Y .33059017 04 Z -.24648340 05 DX -.30553899 01 DY -.40229929 01 DZ -.45496615 01
INC .13464238 03 LAN .21712663 03 APF .23754635 03 MX -.44519377 00 MY -.80995448 00 MZ -.38180656 00
WX .42945024 00 WY .56728643 00 WZ -.70267960 00 PX .78573209 00 PY .14888163 00 PZ .60038754 00
OX -.44519382 00 OY .80995450 00 OZ -.38180645 00 RX .15308932 00 RY .48018912 00 RZ .86730252 00
BX .86414484 00 BY .32097534-01 BZ -.50221853 00 TX .95275257 00 TY .30374750 00 TZ .00000000 00
SX1 -.26234749 00 SY1 .82289480 00 SZ1 .50400192 00 DA1 .30265120 02 RA1 .25231716 03
SX0 .60643402 00 SY0 .75770550 00 SZ0 .24108117 00 DA0 -.13950360 02 RAD .23132775 03
ETE .31365147 03 ETS .40588012 02 ETC .26213010 03

BTC -.39752710 05 BRC .28412022 05 B .48862266 05 THA .14444588 03 T VECTOR IN ECLIPTIC PLANE

X .55244946 03 Y -.32314327 05 Z -.22054476 05 DX -.60668877 01 DY .15145495 01 DZ -.2660164 01
INC .13723007 03 LAN .13851527 03 APF -.23438645 03 MX -.89300706 00 MY .24318280 00 MZ -.37888185 00
WX .44982048 00 WY .50870296 00 WZ -.73408640 00 PX .40908606-01 PY .83280846 00 PZ .55204755 00
OX .89218161 00 OY .21829185 00 OZ .39542464 00 RX .50638496 00 RY .18010642-01 RZ .86211938 00
BX .23526730 00 BY .86039658 00 BZ .45206972 00 TX .3544620-01 TY .99936808 00 TZ .00000000 00
SX1 .86157462 00 SY1 .30643707-01 SZ1 .50670516 00 DA1 .30444610 02 RA1 .17796301 03
SX0 .87948925 00 SY0 .39543587 00 SZ0 .26495338 00 DA0 -.15364182 02 RAD .15579524 03
ETE .30926021 03 ETS .61081140 02 ETC .27339348 03

BTD -.41605749 05 BRD .25621916 05 B .48862266 05 THA .14837414 03 T VECTOR IN ORBIT PLANE OF TARGET

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632427135666	631706661666	630437201056	603730124337	604531616254	603532066024	EARTH
	21174200096		17448304640		000000000000	INITIAL
217727515530	214622511174	617601553043	602611205256	603400721472	603442741230	VENUS
				235605635567	202667111775	END

U MATRIX FOR MAPPING FORWARD				ITERATION NUMBER 0				
X	Y	Z	DX	DY	DZ	HV	AU	
X	.10966738 01	.22628337 00	-.95824143-01	.38971570-04	-.29859219-04	-.13719305-04	.00000000 00	.00000000 00
Y	.26846167 00	.10270700 01	.15922670 00	-.38408640-04	-.12657212-04	-.40772528-05	.00000000 00	.00000000 00
Z	.14890519 00	.18058750 00	-.92539933 00	-.46092930-07	-.31993660-05	-.22687172-04	.00000000 00	.00000000 00
DX	.75527639 06	.15126123 06	.61127321 05	-.27719612 02	-.20412752 02	-.94570449 01	.00000000 00	.00000000 00
DY	.16940998 06	.67661268 06	.87107397 05	-.25112386 02	-.75215715 01	-.23777418 01	.00000000 00	.00000000 00
DZ	.90707747 05	.10145706 06	-.62498159 06	-.10337408 01	-.21220986 01	-.14140570 02	.00000000 00	.00000000 00
HV	-.11663469 11	-.12298978 11	-.11436336 11	-.11459670 07	-.46888873 06	-.27653503 06	.10000000 01	.00000000 00
AU	.10846571 00	.76123472-01	.54280288-01	.20273964-04	.53912774-05	-.67916350-05	.00000000 00	.10000000 01

COVARIANCE MATRIX AT IMPACT				ITERATION NUMBER 0				
X	Y	Z	DX	DY	DZ	HV	AU	
X	.47125286 11	.38625947 11	.20113070 11	.24072435 07	.73161593 06	-.88647863 06	-.15561391-07	.34610615 01
Y	.38625947 11	.34565694 11	.17722058 11	.20695190 07	.54127742 06	-.74744181 06	-.18174269-07	.31844926 01
Z	.20113070 11	.17722058 11	.10110164 11	.10488527 07	.28992393 06	-.41058659 06	-.17080953-07	.22685311 01
DX	.24072435 07	.20695190 07	.10488527 07	.12505391 03	.35761035 02	-.45555512 02	-.13852334-11	.55924926-03
DY	.73161593 06	.54127742 06	.28992393 06	.35761035 02	.12467717 02	-.13385001 02	-.48508475-12	.13441655-03
DZ	-.88647863 06	-.74744181 06	-.41058659 06	-.45555512 02	-.13385001 02	.17365259 02	.43232970-12	-.19567239-03
HV	-.15561391-07	-.18174269-07	-.17080953-07	-.13852334-11	-.48508475-12	.43232970-12	.99999998-18	.00000000 00
AU	.34610615 01	.31844926 01	.22685311 01	.55924926-03	.13441655-03	-.19567239-03	.00000000 00	.25000000 02

U PRODUCT MATRIX OF STEP MAPPING				ITERATION NUMBER 0				
X	Y	Z	DX	DY	DZ	HV	AU	
X	.56437986 01	.42107851 01	.21757652 01	.27753481-03	.95502073-04	-.10171062-03	.00000000 00	.00000000 00
Y	.42107851 01	.48017091 00	.36603342 00	.32273659-04	.12672889-04	-.14935375-04	.00000000 00	.00000000 00
Z	.21757652 01	.36603342 00	.19477862 00	.15866207-04	.21757187-05	.23402609-05	.00000000 00	.00000000 00
DX	.27753481-03	.32273659-04	.15866207-04	.15866207-04	.34895230 03	-.34874327 03	.00000000 00	.00000000 00
DY	.19477862 00	.15866207-04	.15866207-04	.15866207-04	.27889194 02	-.16720472 03	.00000000 00	.00000000 00
DZ	.34895230 03	-.34874327 03	.00000000 00	.00000000 00	.48287167 02	-.15515567 03	.00000000 00	.00000000 00
HV	-.15561391-07	-.18174269-07	-.17080953-07	-.13852334-11	-.48508475-12	.43232970-12	.99999998-18	.00000000 00
AU	.34610615 01	.31844926 01	.22685311 01	.55924926-03	.13441655-03	-.19567239-03	.00000000 00	.25000000 02

ENCOUNTER PARAMETERS AND STATISTICS 62/12/14 194658.745

B	.48863381 05	SHAA	.13827722 06
B.NO	.25623516 05	SHIA	.46249093 05
B.TO	-.41606075 05	DEL T	.31379607 05
B.RC	.28412022 05	DEL B	.14581931 06
B.TC	-.39752716 05	DEL S	.17077879 06
TL	.10077253 03	DEL DR	.92950071 05
THETA	.37570843 02	DEL SF	.11317444 06

N MATRIX

	B-RD	B-TO	TL	C3	S-TS	S-RS
B-RD	.84548154 10	-.82051087 10	.26191344 10	-.59891937 05	.22538312 06	-.11646524 06
B-TO	-.82051090 10	.12808455 11	-.30171756 10	-.64743844 05	-.28456252 06	-.14590043 06
TL	.26191345 10	-.30171755 10	.98467973 09	-.20929726 05	.80506390 05	.41782426 05
C3	-.59891386 05	.64743775 05	-.20929769 05	.45701790 00	-.17441461 01	-.90371404 00
S-TS	.22538311 06	-.28456251 06	.80506391 05	-.17441487 01	.70437215 01	.36340356 01
S-RS	.11646524 06	-.14590043 06	-.41782426 05	-.90371227 00	.36340358 01	.18758385 01

NONNORMALIZED N MATRIX

	B-RD	B-TO	TL	C3	S-TS	S-RS
B-RD	.10000000 01	-.78846768 00	.90773319 00	-.96349549 00	.92356662 00	-.92479722 00
B-TO	-.78846771 00	.99999999 00	-.84958083 00	.84622042 00	-.94738878 00	-.99126186 00
TL	.90777323 00	-.84958081 00	.10000000 01	-.98661894 00	.96657782 00	.97218390 00
C3	-.96348663 00	.84622192 00	-.98662093 00	.10000000 01	-.97219049 00	-.97603794 00
S-TS	.92356665 00	-.94738875 00	.96657784 00	-.97211093 00	.10000000 01	.99974760 00
S-RS	.92479726 00	-.94126187 00	.97218390 00	-.97603603 00	.99974765 00	.99999999 00

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DM/DOO MATRIX

	B.RD	B.TD	TL	C3	S.TS	S.RS
X	.19432538 01	-.30687803 01	.71810839 00	-.15502912-04	.67870250-04	-.34794829-04
Y	.12661832 00	-.37872949 00	-.99066247-01	-.15491778-05	.79940872-05	-.41622495-05
Z	.34317390 00	-.17255082 00	-.66687256-02	.86614341-06	.37648641-05	-.17358850-05
OX	.64180771 07	-.11189817 08	-.23966677 07	-.51176269 02	.23588366 03	-.12053721 03
OY	.65818714 07	-.15729188 07	.16748565 07	-.41920920 02	-.11365567 03	.60119785 02
OZ	-.18816308 06	-.63123703 05	.11391386 07	-.15014267 02	-.43341703 02	-.24794701 02
HU	-.56560495-01	-.96096551-01	-.26623533-01	.14224868-03	.50777522-06	-.40448822-06

DM/DO MATRIX

	B.RD	B.TD	TL	C3	S.TS	S.RS
X	-.21496986 00	-.71465895 00	.55341266-01	.33640446-03	-.10118067-04	-.62313123-05
Y	.60157931 00	.64188397 00	.14796535 00	-.11025433-03	.20058079-05	-.12722463-05
Z	-.91472661 00	-.98744307-01	.97671857-01	-.23437026-03	-.56003876-05	-.51622475-06
OX	-.11462816 04	-.42096133 04	-.67860414 03	-.71121789 01	-.20354437-01	-.12535791-01
OY	.42021899 04	-.25239673 04	.46801974 03	-.10264800 02	-.17521200-01	-.10790601-01
OZ	-.82683492 04	-.68900916 03	.67421384 03	-.53769287 01	-.57946112-01	-.42589667-01
HU	-.33843363-01	-.14002395 00	-.17583676-02	.91201498-04	.18249393-05	-.11239068-05

N MATRIX FOR MANEUVER CALCULATIONS

	B.RC	B.TC	TL
B.RC	.95964167 10	-.84255963 10	-.32634384 05
B.TC	-.84255966 10	.11666853 11	-.32763030 05
TL	.32634384 05	-.32763030 05	.13190690 00

0000 141631 OFFLINE CONTROL
 0000 14163 TERMINATE JOB
 0000 14163 END DATA

127
 10

SOURCE PROGRAM LISTING

```

*H1  QDPX  Y9
*H9  L49  X,N11,N1,N4
55164/1.0E14
*H12 L412 X
*H13  SPACE X
* SCDFRF=2
* WANT  N9,(MDDPH1),Z0D,(INDTOR,BRDPT),18
* WANT  N11,(TAPEX),6
* WANT  N12,(XNJ,INJ),1,(XNJ-1,INJ),1
* ETC   (XNJ-2,INJ),1,(XNJ-3,INJ),1
* ETC   (XNJ-4,INJ),1,(XNJ-5,INJ),1
* ETC   (EPDCH,INJ),1,(EPDCH-1,INJ),1,1
* ETC   (DUP,GRAY-2),1,(DUP,SCALE),1,1
* ETC   (RE,SCALE),1,(PORE,HARMN4),1
* ETC   (QU,HARMN4),1,(QU,HARMN3),1
* ETC   (QJ,HARMN2),1,(GRAVE,GRAY),1
* ETC   (GRAV,GRAY+1),1,(GRAV,GRAY+2),1
* ETC   (GRAV,GRAY+3),1,(GRAV,GRAY+4),1
* ETC   (CHAV,GRAY+6),1,(STEP,H),1
* ETC   (TANGFT,FARBCO),1,(AREA,RADDP),1
* ETC   (GM0,RADDP),1,(FMAS,RADDP),1
* ETC   (FAD,GASDP),1,(FBD,GASDP),1
* ETC   (FC0,GASDP),1,(FA1,GASDP),1
* ETC   (FB1,GASDP),1,(FC1,GASDP),1
* ETC   (FA2,GASDP),1,(FB2,GASDP),1
* ETC   (FC2,GASDP),1,(JETON,GASDP),1
* ETC   (JETON,GASDP),1,(JETON-1,GASDP),1
* ETC   (JETOFF,GASDP),1,(JETOFF-1,GASDP),1
* ETC   (IF1,SFCP50),1,(IFIN-1,SECP50),1
* ETC   (FMAS,GASDP),1
GASDP=(CANOPU)
INJCO=(EARTH)
*H11  L411  Y4,N6,N12
* WANT  Y9,(C137),1
* WANT  N13,(CLPT,TRADE),1,(CLPT,TRADE),1
* ETC   (CLPTC,TRADE),1,(CLPBC,TRADE),1
* ETC   (CLPBT,TRADE),1,(CLPBT,TRADE),1
* ETC   (TAPEX),6
*H2  L42  N12,X
* WANT  N11,(EPOX),14
*H2A  L42A  Y9,N12
*H3  L43  Y9,N10
*H4  L44  Y6,N12
*H5  L45  Y9,N12
*H6  L46  N12
*H7  L47  Y9,N10,N12
* WANT  Y9,(APR1DR-19,APR1DR-19),20
*H8  L48  Y9,N12
*H10  L410  Y8,Y9
*H  END
    
```

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THERE WERE NO CLARIFY SOURCE DECK ERRORS.
 THE OBJECT STRING HAS 00514 OCTAL OR 332 DECIMAL WORDS.

JPL TECHNICAL MEMORANDUM NO. 33-204

```

0000 141751 PAGE HEADING (23)
0000 14175 (MOON CHECK CASE - PREDICTIONS)
0000 14175 EPDCH (101)
0000 14175 640702910,2758000
0000 14175 GEOCENTRIC POSITION AND VELOCITY AT EPDCH (102)
0000 14175 X=.15667453E5 Y=.63041615E5 Z=.80777204E4
0000 14175 DX=.14342616E1 DY=.97256996E2 DZ=.28116199E0
0000 14175 OTHER OPTIONS AND CONSTANTS (103)
0000 14175 SYNFRQ=2038516.
0000 14175 TFREQ=7590330.
0000 14175 POINTING TIMES, SAMPLE RATE, COUNT TIMES (117)
0000 14175 JEIDL2=640702911,0,640703011,0,1800.,.60.
0000 14175 TRANSMITTER ID TABLE (126)
0000 14175 2,2038516.
0000 14175 OFFLINE CONTROL (127)
0000 14175 KEY(8) KEY(13)
0000 14175 END DATA (10)
    
```

CASE 1 IBSYS-JPTRAJ-SPACE 022265 1

DOUBLE PRECISION EPHEMERIS TAPE - EPHEM1

```

GME .39860063 06 J .16234500-02 H -.57499999-05 D .78749999-05 RE .63781650 04 REM .63783112 04
G .66709998-19 A .88781796 29 B .88800194 29 C .88836976 29 DME .41780741-02 AU .14959850 09
GMW .49026293 04 GMS .13271411 12 GMV .32476627 05 GMA .42977368 05 GNC .37918700 08 GHJ .12670935 09
EGM .39860320 06 MGH .49027779 04 JA .29200000-02 HA .00000000 00 DA .00000000 00 RA .34170000 04
    
```

INJECTION CONDITIONS MOON 235666506353202400000000 J.D.= 2438605.93608796 JULY 29, 1964 10 27 58.000

```

GEOCENTRIC X0 .15667453 06 Y0 .63041615 05 Z0 .80777203 04 DX0 .14342616 01 DY0 .97256996 00 DZ0 .28116199 00
CARTESIAN ID .37678000 05 GHA .10409373 03 GHD .30667227 03
DATE OF RUN 022265A 14185 EARTH IS THE CENTRAL BODY FOR INTEGRATION COMELL EQUATIONS OF MOTION
    
```

0 DAYS 0 HRS. 0 MIN. 0.000 SEC. 235666506353202400000000 J.D.= 2438605.93608796 JULY 29, 1964 10 27 58.000

GEOCENTRIC EQUATORIAL COORDINATES

```

X .15667452 06 Y .63041612 05 Z .80777202 04 DX .14342615 01 DY .97256992 00 DZ .28116198 00
R .16907513 06 DEC .27384004 01 RA .21918529 02 V .17555769 01 PTH .76231923 02 AZ .61412202 02
B .16907512 06 LAT .27384004 01 LON .27782479 03 VE .12070910 02 PTE .81207508 01 AZE .27095862 03
XS .89949372 08 YS .11227336 09 ZS .48686598 08 DXS -.23515989 02 DYS -.16077681 02 DZS -.69719988 01
XM .38246410 06 YM .27456507 05 ZM -.26012551 05 DXM -.83439888-01 DYM .93230192 00 DZM .40985490 00
XT .38246410 06 YT .27456507 05 ZT -.26012551 05 DXT -.83439888-01 DYT .93230192 00 DZT .40985490 00
RS .15187686 09 VS .29327501 02 RM .38432968 06 VM .10218268 01 RT .38432968 06 VT .10218268 01
GED .27570333 01 ALT .16626784 06 LOS .24408715 02 RAS .12870045 03 RAM .41061295 01 LDM .26001239 03
GUT .35000000 02 DT .12000000 03 DR .17051340 01 SHA .16335718 06 DES .18697172 02 DEH -.38809109 01
CCL .25840724 03 MCL .11052996 00 TCL .11052996 00
    
```

GEOCENTRIC CONIC

```

EPOCH OF PERICENTER PASSAGE 23566644500622026213000000 J.D.= 2438605.21642518 JULY 28, 1964 17 11 39.136
SHA .24408791 02 ECC .97401695 00 B .55279828 05 SLR .12519505 05 APD .48183367 06 RCA .63421468 04
YH .14661061 00 C3 -.16330208 01 CI .70641933 05 TFP .62178864 05 TF -.17271907 02 PER .20002258 05
TA .16192549 03 NTA .18000000 03 EA .71607988 02 MA .18651553 02 C3J -.20370820 01 TFI .00000000 00
    
```

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE

```

X .15667452 06 Y .63041612 05 Z .80777202 04 DX .14342615 01 DY .97256992 00 DZ .28116198 00
INC .28976671 02 LAN .17015325 02 APF .20329595 03 MX -.45905374 00 MY .75220666 00 MZ .47271026 00
NX .14176438 00 NY -.63274717 00 WZ .87481701 00 PX -.77702823 00 PY -.55959958 00 PZ .19151913 00
QX .61926389 00 QY -.65062065 00 QZ -.43955031 00 RX .15255788 00 RY .11936638 00 RZ .98147647 00
SX .61926406 00 SY .65062083 00 SZ .43955042 00 TX -.61622264 00 TY .79169463 00 TZ .00000000 00
DAP -.11169176 02 RAP .21765584 03
BTQ .49421011 05 BRQ -.24767379 05 B .55279828 05 THA .33338222 03 T VECTOR IN EARTH EQUATOR PLANE
    
```

CASE 1 IBSYS-JPTRAJ-SPACE 022265 2

HELIOCENTRIC EQUATORIAL COORDINATES

```

X .90106046 08 Y -.11221032 09 Z -.48678520 08 DX .24950250 02 DY .17050251 02 DZ .72531608 01
R .15192055 09 LAT .18688381 02 LON .30876483 03 V .31077876 02 PTH .21988085 00 AZ .75813410 02
XE .89949372 08 YE -.11227336 09 ZE -.48686598 08 DXE .23515989 02 DYE .16077681 02 DZE .69719988 01
XT .90331836 08 YT .11224590 09 ZT -.48712610 08 DXT .23432549 02 DYT .17009983 02 DZT .73818538 01
LTE .18697172 02 LDE .30870045 03 LTT .18680121 02 LDT .30882598 03 DST .15209175 09 YST .29881694 02
EPS .74994993 02 ESP .61373100-01 SEP .10494339 03 SEH .14723361 03 ESH .13773981 02 MEP .18992403 02
MPS .13777126 03 MSP .59333450-01 SMP .42170224 02 SEM .12399574 03 EMS .95944139 02 ESM .12012787 00
RPM .23110470 06 SPX .72833120 02 ZT .10548403 05 DXT .32284243 00 DYT .88717703 00 DZT .41420339 00
RST .15189594 09 YS .29333497 02 RM .38036037 06 VM .10309582 01 RT .38036037 06 VT .10309582 01
GCE .10159276 03 GCT .28170329 03 SIP .13734034 03 CPT .92025110 02 SIN .91594195 02
REP .16907513 06 VEP .17555769 01 CPE .97484351 02 CPS .76877815 02
    
```

1 DAYS 0 HRS. 32 MIN. 2.000 SEC. 235666561454202000000000 J.D.= 2438606.95833333 JULY 30, 1964 11 00 00.000

GEOCENTRIC EQUATORIAL COORDINATES

```

X .25084050 06 Y .13402614 06 Z .30322890 05 DX .81614400 00 DY .67379618 00 DZ .22454298 00
R .28601301 06 DEC .60858957 01 RA .28115954 02 V .10819020 01 PTH .77274393 02 AZ .61615284 02
R .28601300 06 LAT .60858957 01 LON .27500634 03 VE .20556600 02 PTE .29427204 01 AZE .27031620 03
XS .92013117 08 YS .11083678 09 ZS .48063650 08 DXS -.23214413 02 DYS -.16451548 02 DZS .71333863 01
XM .36449540 06 YM .10819357 06 ZM .10548403 05 DXM .32284243 00 DYM .88717703 00 DZM .41420339 00
XT .36449540 06 YT .10819357 06 ZT .10548403 05 DXT .32284243 00 DYT .88717703 00 DZT .41420339 00
RS .15189594 09 VS .29333497 02 RM .38036037 06 VM .10309582 01 RT .38036037 06 VT .10309582 01
GED .61270553 01 ALT .27963504 06 LOS .16588743 02 RAS .12969836 03 RAM .16532524 02 LDM .26342290 03
GUT .35000000 02 DT .48000000 03 DR .10953263 01 SHA .28251917 06 DES .18451435 02 DEM .15891664 01
CCL .25920166 03 MCL .11192516 01 TCL .11192516 01
    
```

GEOCENTRIC CONIC

```

EPOCH OF PERICENTER PASSAGE 23566645033320254440000000 J.D.= 2438605.22424520 JULY 28, 1964 17 22 54.785
SHA .24685405 06 ECC .97407359 00 B .53607772 05 SLR .11656495 05 APD .48718128 06 RCA .58988163 04
YH .13991439 08 C3 .16161877 01 C1 .68163673 05 TFP .14982521 06 TF -.17084226 02 PER .20304432 05
TA .16934513 03 NTA .18000000 03 EA .99441039 02 MA .44273647 02 C3J .20471674 01 TFI .24533889 02
    
```

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE

```

X .25084050 06 Y .13402614 06 Z .30322890 05 DX .81614400 00 DY .67379618 00 DZ .22454298 00
INC .28976671 02 LAN .17015325 02 APF .20329595 03 MX -.45905374 00 MY .75220666 00 MZ .47271026 00
NX .14176438 00 NY -.63274717 00 WZ .87481701 00 PX -.77702823 00 PY -.55959958 00 PZ .19151913 00
QX .61926389 00 QY -.65259660 00 QZ -.44495801 00 RX .15168230 00 RY .11704677 00 RZ .98147647 00
SX .61329438 00 SY .65259664 00 SZ .44495801 00 TX .61091702 00 TY .79169463 00 TZ .00000000 00
DAP -.11045703 02 RAP .21765584 03
BTQ .47782174 05 BRQ -.24303436 05 B .53607772 05 THA .33304076 03 T VECTOR IN EARTH EQUATOR PLANE
    
```

JPL TECHNICAL MEMORANDUM NO. 33-204

CASE 1 18SYS-JPTRAJ-SPACE 022265 3

HELIOCENTRIC EQUATORIAL COORDINATES

X -92263957 08	Y -11070276 09	Z -48033337 08	DX -24030557 02	DY -17125344 02	DZ -73583793 01
R -15190437 09	LAT -18433738 02	LDN -30980913 03	V -30412018 02	PTH -39832041 00	AZ -75360793 02
XE -92013117 08	YE -11083678 09	ZE -48063660 08	DXE -23214413 02	DYE -16451548 02	DZE -71338363 01
XI -92377612 08	YI -11072859 09	ZI -48053111 08	DXT -22891571 02	DYI -17335725 02	DZI -75480396 01
LIE -18451435 02	LOE -30969936 03	LIT -18429768 02	LOT -30983725 03	LST -15199850 09	VST -29692226 02
EPS -80928671 02	ESP -10653118 00	SEP -98964764 02	EPH -13630553 03	EMP -31295930 02	MEP -12399535 02
MPS -14275634 03	MSP -25217635-01	SMP -37216690 02	SEM -11136139 03	EMS -68505077 02	ESH -13343921 00
RPM -11821424 06	SPV -79650888 02				
GCE -10079833 03	GCT -28191758 03	SIP -14191393 03	CPT -93302061 02	SIN -92459653 02	
REP -28601301 06	VEP -10819020 01	CPE -98321564 02	CPS -76978757 02		

1 DAYS 0 HRS. 36 MIN. 7.000 SEC. 235666561550202000000000 J.D.= 2438606.96111111 JULY 30,1964 11 04 00.000

GEOCENTRIC EQUATORIAL COORDINATES

X -25103626 06	Y -13418778 06	Z -30376764 05	DX -81519112 00	DY -67322571 00	DZ -22440404 00
R -28626616 06	DEC -60913377 01	RA -28126092 02	V -10807990 01	PTH -77275751 02	AZ -61614519 02
XE -28626615 06	LAT -60913377 01	LDN -27401373 03	VE -20574902 02	PTE -29371153 01	AZE -27031557 03
XI -92018688 08	YI -11083283 09	ZI -48061947 08	DXI -23213584 02	DYI -16452559 02	DZI -71342736 01
XM -36441784 06	YM -10840647 06	ZM -10647809 05	DXM -32348844 00	MYM -88698269 00	DZM -41418345 00
XI -36441784 06	YI -10840647 06	ZI -10647809 05	OXT -32348844 00	OYT -88698269 00	OZI -41418345 00
RS -15189590 09	VS -29333513 02	RM -38034944 06	VM -10309843 01	RI -38034944 06	VI -10309843 01
GEU -61325336 01	ALT -27988819 06	LOS -15588715 02	RAS -12970107 03	RAM -16566608 02	LOM -26245425 03
DUT -33000000 02	DT -48000000 03	DR -10542562 01	SHA -28277598 06	UES -18450759 02	DEH -16041929 01
CCL -25920271 03	MCL -11199342 01	TCL -11199342 01			

GEOCENTRIC COMIC

EPDM OF PERICENTER PASSAGE 2356664503352020060000000 J.D.= 2438605.22430609 JULY 28,1964 17 23 00.047

SMA -24655219 06	ECC -97608645 00	B -53596202 05	SLR -11650884 05	APD -48720843 06	RCA -58959386 04
VA -13987292 00	C3 -16166988 01	C1 -68147266 05	IFP -15005995 06	IF -17082765 02	PER -20305931 05
TH -16935879 03	MTA -18000000 03	EA -99498608 02	MA -44339739 02	C3J -120471971 01	TFI -24600555 02

ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE

X -25103626 06	Y -13418778 06	Z -30376764 05	DX -81519112 00	DY -67322571 00	DZ -22440404 00
INC -28978429 02	LAM -17016156 02	APP -20329300 03	HX -45922455 00	MY -75209822 00	MZ -47271731 00
HX -14177917 00	HY -46327196 00	WZ -87480215 00	PX -77705276 00	PY -59957162 00	PZ -19154992 00
QX -61325999 00	QY -65260538 00	QZ -64449313 00	KK -15167722 00	RY -11703370 00	RZ -98147700 00
BX -61325984 00	BY -65260523 00	BZ -44449303 00	TX -61088707 00	FY -79171775 00	FZ -00000000 00
OAP -11045001 02	RAP -21765367 03				

BTQ -47770919 05 BRQ -24300045 05 B -53596202 05 THA -33303853 03 T VECTOR IN EARTH EQUATOR PLANE

CASE 1 18SYS-JPTRAJ-SPACE 022265

HELIOCENTRIC EQUATORIAL COORDINATES

X -92269724 08	Y -11069865 09	Z -48031570 08	DX -24028775 02	DY -17125784 02	DZ -73586777 01
R -15190432 09	LAT -18433041 02	LDN -30981194 03	V -30410929 02	PTH -39840322 00	AZ -75359724 02
XE -92018688 08	YE -11083283 09	ZE -48061947 08	DXE -23213584 02	DYE -16452559 02	DZE -71342736 01
XI -92383105 08	YI -11072443 09	ZI -48051299 08	DXT -22890099 02	DYI -17339541 02	DZI -75480471 01
LIE -18450759 02	LOE -30970107 03	LIT -18429081 02	LOT -30983999 03	LST -15199824 09	VST -29691674 02
EPS -80937248 02	ESP -10653118 00	SEP -98956090 02	EPH -13628907 03	EMP -31338244 02	MEP -12372679 02
MPS -14276421 03	MSP -25217635-01	SMP -37208088 02	SEM -11132686 03	EMS -68539587 02	ESH -13343921 00
RPM -11793766 06	SPV -79660594 02				
GCE -10079728 03	GCT -28191721 03	SIP -14191979 03	CPT -93304393 02	SIN -92459974 02	
REP -28626616 06	VEP -10807990 01	CPE -98322790 02	CPS -76979031 02		

222462325467 220750470430 215753222173 201560037436 200757472061 177435677052 EARTH INITIAL

21187169515 17515413504 000000000000

222753160271 222404413161 217725145467 200642503745 200527270151 176711206656 EARTH END

235666561550 202000000000

JPL TECHNICAL MEMORANDUM NO. 33-204

TIME	STATION NUMBER 2			64/07/29					PAGE 1			
	R	DR	EL	AZ	DEC	HA	CI	CC3		E3	RU	
110000	-16755796	06	-15050596	01	47.636	136.455	1.705	332.326	1010789.13	1023069.11	1023069.11	0011442252115
113000	-17028671	06	-15274765	01	51.590	146.347	1.846	339.850	1010960.92	1023412.66	1023412.66	0011562336493
120000	-17305866	06	-15529082	01	54.596	157.866	1.982	347.384	1011155.81	1023807.44	1023802.44	0011703362333
123000	-17587845	06	-15804852	01	56.389	170.809	2.113	354.924	1011367.16	1024225.10	1024225.10	0012025722225
130000	-181874913	06	-16093108	01	56.767	184.497	2.238	2.466	1011588.06	1024666.91	1024666.91	0012151453753
133000	-18167214	06	-16384802	01	55.686	197.936	2.358	10.006	1011811.61	1025114.00	1025114.00	0012277222501
140000	-18464729	06	-16670989	01	53.281	210.238	2.473	17.540	1012030.93	1025552.64	1025552.64	001242621237
143000	-18767281	06	-16943014	01	49.800	220.972	2.584	25.066	1012239.39	1025969.58	1025969.58	0012556551663
150000	-19074539	06	-17192679	01	45.509	230.131	2.690	37.500	1012430.72	1026352.23	1026352.23	0012710431953
153000	-19386432	06	-17442398	01	40.635	231.930	2.791	50.079	1012599.08	1026688.97	1026688.97	001304347341
160000	-19701158	06	-17595327	01	35.353	244.648	2.890	47.561	1012739.23	1026969.29	1026969.29	001317757503
163000	-20019198	06	-17735484	01	29.790	250.542	2.985	55.024	1012846.56	1027183.99	1027183.99	001333456273
170000	-20339335	06	-17827840	01	24.042	255.803	3.078	62.466	1012917.20	1027325.29	1027325.29	001347224263
173000	-20660669	06	-17868442	01	18.179	260.665	3.171	69.884	1012948.01	1027366.95	1027366.95	001362020561
180000	-20982235	06	-17854687	01	12.248	265.250	3.261	77.281	1012936.64	1027386.27	1027386.27	001376621393
183000	-21303024	06	-17787053	01	6.368	269.642	3.392	84.596	1012881.63	1027254.27	1027254.27	001412403511
190000	-21622002	06	-17707772	01	.803	273.967	3.697	91.643	1012782.75	1027056.52	1027056.52	001426123757

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070000	-27016468	06	-74415302	00	1.327	84.522	5.232	267.906	1004988.50	1011468.10	1011468.10	0017333236605
073000	-27150337	06	-73977632	00	7.101	88.762	5.109	275.090	1004928.05	1011347.15	1011347.15	0017401444427
080000	-27283360	06	-73864767	00	13.138	93.055	5.125	282.487	1004916.70	1011324.41	1011324.41	0017450353411
083000	-27416693	06	-74363618	00	19.237	97.503	5.181	289.962	1004954.21	1011399.38	1011399.38	001751734177
090000	-27551468	06	-75475478	00	25.283	102.228	5.234	297.453	1005039.15	1011569.20	1011569.20	0017566657595
093000	-27688772	06	-77173927	00	31.236	107.315	5.289	304.984	1005169.18	1011829.23	1011829.23	0017636276711
100000	-27829430	06	-79419018	00	37.030	113.131	5.344	312.493	1005341.17	1012173.16	1012173.16	001771012373
103000	-27974982	06	-82159878	00	42.574	119.735	5.398	320.037	1005551.19	1012593.16	1012593.16	001776252055
110000	-28125670	06	-85336002	00	47.745	127.498	5.450	327.594	1005794.57	1013079.91	1013079.91	002003655402
0000	142044		POINTING TIMES									(17
0000	14204		JETOLV640702911,0,640703011,0,1800,60,									
0000	14204		TRANSMITTER IO TABLE									(26
0000	14204		1,2038516,									
0000	14204		OFFLINE CONTROL									(27
0000	14204		KEY(13)									
0000	14204		END DATA									10

TIME	STATION NUMBER 1			64/07/29					PAGE 1			
	R	DR	EL	AZ	DEC	HA	CI	CC3		E3	RU	
110000	-16754786	06	-19550839	01	47.774	136.409	1.709	332.381	1010789.32	1023069.48	1023069.48	0011444227211
113000	-17027675	06	-15274002	01	51.738	146.322	1.851	339.905	1010961.87	1023414.55	1023414.55	0011562311137
120000	-17304902	06	-15531298	01	54.750	157.873	1.986	347.439	1011157.52	1023805.83	1023805.83	001170314205
123000	-17586929	06	-15808013	01	56.542	170.878	2.117	354.980	1011369.58	1024229.95	1024229.95	0012025511115
130000	-17874062	06	-16097161	01	56.911	184.619	2.242	2.522	1011591.17	1024673.13	1024673.13	0012151454445
133000	-18166443	06	-16389676	01	55.813	198.099	2.362	10.062	1011815.34	1025121.47	1025121.47	001227704147
140000	-18464053	06	-16676600	01	52.386	210.424	2.477	17.597	1012035.23	1025541.24	1025541.24	001242604517
143000	-18766711	06	-16949262	01	49.891	221.162	2.587	25.123	1012244.18	1025979.16	1025979.16	001255643443
150000	-19074087	06	-17194454	01	45.568	230.310	2.693	32.637	1012435.91	1026362.62	1026362.62	001271032575
153000	-19385706	06	-17419581	01	40.674	238.093	2.795	40.136	1012604.59	1026699.98	1026699.98	001304341251
160000	-19700963	06	-17802795	01	35.374	244.791	2.893	47.614	1012744.95	1026980.73	1026980.73	001317753775
163000	-20019140	06	-17743108	01	29.796	250.665	2.988	55.082	1012852.61	1027195.68	1027195.68	001333455217
170000	-20339415	06	-17835493	01	24.034	255.933	3.081	62.524	1012923.06	1027337.02	1027337.02	001347225653
173000	-20660885	06	-17875998	01	18.158	260.770	3.174	69.941	1012953.80	1027398.52	1027398.52	001363024613
180000	-20982585	06	-17862029	01	12.216	265.317	3.264	77.338	1012942.27	1027375.51	1027375.51	001376267777
183000	-21303504	06	-17794119	01	6.325	269.694	3.396	84.652	1012886.99	1027264.98	1027264.98	001412414471
190000	-21622694	06	-17716328	01	.759	274.003	3.705	91.691	1012787.78	1027066.60	1027066.60	001426137161

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070000	-27016126	06	-74354145	00	1.353	84.548	5.231	267.956	1004983.37	1011457.84	1011457.84	001733225267
073000	-27149878	06	-73919059	00	7.141	88.773	5.111	275.141	1004923.27	1011337.59	1011337.59	001740132761
080000	-27282795	06	-73808231	00	13.190	93.050	5.128	282.540	1004912.35	1011315.72	1011315.72	001745022795
083000	-27416031	06	-74313785	00	19.300	97.484	5.184	290.015	1004950.39	1011391.74	1011391.74	001751717675
090000	-27550723	06	-75433312	00	25.358	102.194	5.237	297.506	1005035.91	1011562.73	1011562.73	001756650015
093000	-27687959	06	-77140234	00	31.322	107.328	5.292	305.017	1005166.59	1011824.06	1011824.06	001763657613
100000	-27828764	06	-79394454	00	37.128	113.072	5.347	312.547	1005339.29	1012169.40	1012169.40	001770722225
103000	-27974081	06	-82144929	00	42.686	119.669	5.400	320.091	1005550.04	1012590.87	1012590.87	001776231171
110000	-28124751	06	-85330990	00	47.870	127.430	5.453	327.648	1005794.19	1013079.13	1013079.13	002003634236
0000	14211		OFFLINE CONTROL									(27
0000	14211		TERMINATE JOB									
0000	14211		END DATA									10

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