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September 1, 19<sup>th</sup>

National Aeronautics & Space Administration  
Electronic Research Center  
575 Technology Square  
Cambridge, Massachusetts 02139

Attn: Computer & Guidance Technology Division

Dear Sir:

Enclosed are one reproducible (and two duplicated copies) of the final technical report, "Designer's Manual for Computer-Aided Design Control Circuits", prepared by S. D. Bedrosian and D. I. Howe under Contract NAS12-2137.

This completes the report requirements for the subject contract.

Sincerely,

  
Edward J. Parker

cc: Dr. S. D. Bedrosian  
Mr. R. L. Keane, ONR Resident Representative  
Mr. A. Merritt, ORA  
Contract file (MS6920)

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DESIGNERS MANUAL FOR COMPUTER-AIDED

DESIGN OF CONTROL CIRCUITS

FINAL TECHNICAL REPORT

Contract NAS 12-2137

National Aeronautics and Space Administration  
Electronics Research Center  
575 Technology Square  
Cambridge, Massachusetts 02139

Prepared by

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June 15, 1970

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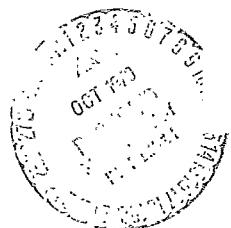


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CHAPTER I  
INTRODUCTION

This manual describes the necessary preliminary electric circuit modeling and methods of computer analysis using the NASAP digital computer program to aid in the design of control circuits used in aerospace systems. The scope of design treated is limited to single-input single-output linear time invariant control systems that can be described by rational transfer functions.

Two different approaches can be followed in the design of such linear time invariant feedback control systems. In one case the designer seeks compensators for a given plant to satisfy the over-all system requirements. This approach generally involves cut and try and the well known conventional design techniques such as the Bode and Nyquist plots and use of the root locus. In the other case, the designer starts by obtaining the over-all transfer function from the given plant and the specifications. He is then in a position to determine the required compensators. The second of these two approaches seems more amenable to computer-aided design of control circuits using the NASAP program.

NASAP is an acronym for Network Analysis for Systems Applications Program. NASAP has been developed and is being maintained by the Automated Techniques Branch, NASA Electronics Research Center, Cambridge, Massachusetts. The NASAP program is based on Mason's signal-flow graph [MA 1] as extended by Happ for the closed signal-flow graph [HA 1]. The program is based upon symbol oriented techniques which with proper tagging and loop evaluation permit both the transfer function and the sensitivity to be made available.

This chapter provides brief discussions on the basic options presently available in the NASAP program and the basic procedures used in computer aided design of control circuits.

## The NASAP Program

NASAP is a linear electrical circuit analysis program which computes a specified transfer function in terms of the complex frequency variable,  $s$ . The basic coding rules and description of the program are contained in the book let "Coding Instructions for NASAP 69/I" by Gaertner Research Incorporated [GA 1] written under NASA contract NAS 12-663. A number of special options are introduced and described later in this manual. The input for NASAP is by punched card and the output is by printer tabulation of data and printer graphics. It is worth noting that the input format of NASAP is simple and that use of this program requires little knowledge of circuit theory or computer programming.

The present version of NASAP can handle linear circuits which consist of constant-value passive elements, and independent or dependent current and voltage sources. The dependent sources must be linearly related to a voltage or current in another part of the circuit. Nonlinear functional relationships (dependencies) and time-varying parameters cannot be handled.

Embodyed in NASAP is the ability to give both a mathematical formulation and a numerical tabulation of the output results with some printer graphics.

A brief summary of the options presently available in the program available on the ~~RCA Spectra 70/46~~ at the Moore School, follows:

1. OUTPUT - The transfer function which is specified by the user is printed out as the ratio of two polynomials in the complex frequency variable,  $s$ . The poles and zeros of the transfer function are evaluated.
2. FREQ - A Bode plot of the transfer function is printed in tabular and graphical form.
3. TIME - The impulse response of the network is printed in tabular and graphical form. To facilitate control circuit design we

have also made available the step and ramp responses.

4. SENSITIVITY - The sensitivity of the transfer function is computed with respect to a designated element in the network. Furthermore the program can print out in numerical form: the sensitivity of the real part of the transfer function, of the imaginary part of the transfer function, of the magnitude of the transfer function, and of the phase of the transfer function to changes in specified circuit parameters.

Details of these options are documented in Section ID.

#### Procedures Used in Computer Aided Design

The major steps in computer aided circuit design are:

- a. Choose the electrical circuit model for the control system.
- b. Calculate element values from design equations.
- c. Analyze circuit using computer.
- d. Change element values if computer analysis results in discrepancies between the electrical circuit model response and desired control circuit response.

Before a control system can be analyzed using NASAP and before the necessary compensation can be determined, the plant's dynamic characteristics must be first simulated by an electric network which has an equivalent dynamic characteristic. The choice of circuit models is based on the specified response for the control circuit. Details of circuit modeling are given in chapter III. Computer analysis is amply demonstrated in chapter IV.

## IA. REVIEW OF CODING PROCEDURES FOR NASAP

### Introduction

NASAP is a computer-aided electrical circuit analysis program which can be used by engineers without knowledge of computer programming. Using NASAP to analyze a circuit it is possible to calculate a transfer function, the sensitivity of the transfer function, the frequency response, and the impulse response. The main steps for preparing NASAP computer instructions are:

- a) Obtain an electrical circuit model for the control circuit
- b) Preparation of the circuit diagram in a form from which all the information required by the computer can be readily extracted.
- c) The preparation of the computer instructions themselves.

The first step will be discussed in chapter III. The second step has been described elsewhere [GA 1] and will be illustrated by many examples throughout this manual. The third step will be reviewed in this chapter for the convenience of the reader. For additional information, see [GA 1]. The coding rules will be presented by an example taken virtually unchanged from [MO 1].

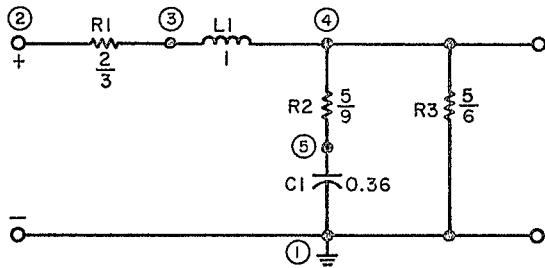
### Example: RLC Filter

We shall analyze the RLC filter shown in Fig. 1.1(a) to find its open circuit transfer voltage function.

### Numbering the Nodes and Elements

The first step in preparing the circuit for NASAP analysis is to number the circuit nodes and elements. The nodes must be numbered sequentially, starting with 1, without skipping any numbers.

The second step is numbering all circuit elements (resistors, capacitors, inductors, and current and voltage sources) consecutively within each category i.e.,  $R_1, R_2, \dots, L_1, L_2, \dots, C_1, C_2$ , etc. No two elements can have the same number designation regardless of their component values. One



```

    → NASAP PROBLEM (RLC FILTER) → COLUMNS
    ↓   ↓   ↓
    V1 1 2 1.0
    L1 3 4 1.0H
    R1 2 3 0.66667
    R2 4 5 0.55556
    C1 5 1 0.36F
    R3 4 1 0.8333
    } NETWORK ELEMENT TABLE

    → OUTPUT
    VR3/VV1
    FREQ -1.2 1.0 0.05
    TIME 5.0
    } OUTPUT SPECIFICATIONS

    → EXECUTION

    CONTROL STATEMENTS      (b)
  
```

Fig. 1.1: RLC Filter Network and the Corresponding NASAP Input Instructions

possible set of number assignments for the RLC filter is shown in Fig. 1.1(b).

#### - Overall Appearance of Input Instructions

Figure 1.2 shows the general appearance of the input instructions. The instructions fall into three different categories: a) Control Statements, b) Network Element Table, and c) Output Specifications. Each type of instructions will be explained briefly below.

#### Control Statements

The first line of the set of instruction must start with the word: NASAP which simply indicates the beginning of a circuit description. The letter N must be in the first column of the card. The word NASAP may be followed by any identifying information but only the first 50 columns of the card will be printed by the computer.

The end of the network element table is designated by the control card with the word: OUTPUT on it. Following the OUTPUT control card are the output specifications. The final control card appearing after the output specifications is the card with the word: EXECUTION.

This card signals the end of the description for the problem and starts the computer on the task of analysis. If it is desired to analyze several problems in one run, the EXECUTION card can be followed by another NASAP card, etc.

#### Network Element Table

##### General Format

The network elements and topology are described in a five column network element table. (Note that the example in Fig. 1.2 has only four columns since no dependent generators are used.) A brief description of each column follows:

Column 1: Identifies the element itself, i.e., V1, R3, etc.

Column 2 and 3: Define the nodes between which the element appears. The positive current direction is always from the first to the second node. Thus, for passive elements the first node is assumed positive with respect to the second, and for active elements the second node is assumed positive with respect to the first.

Column 4: Contains the value of the circuit component.

Column 5: Used only for dependent sources and will be described in detail later.

#### Input Voltage or Current

The first line after the NASAP control statement contains a description of the applied voltage or current. For example, the statement V1 1 2 1.0 in Fig. 1.2 indicates that V1 is a voltage of magnitude 1.0 applied between nodes 1 and 2 with node 2 being positive. The fourth column contains the value of the generated current or voltage and is usually set to 1.0. If any other value is used, it will act as a constant multiplier for the transfer function under consideration.

#### Resistors, Capacitors and Inductors

Resistors, capacitors and inductors are identified in column 1 by the letters R, C, and L, respectively. The letter is immediately followed by a one or two digit integer indicating the element number. The fourth column contains the component value. The value is a number followed by up to two letters which denote a multiplying factor according to the convention given in Table 1.1. The network element table in Fig. 1.2 follows directly from the circuit diagram of Fig. 1.1 (b).

Table E.1. Units Following the Component Values

Letters Used in NASAP	Electrical Units	Multiplying Factor
No letter	Ohms	1
K	Kilohms	$10^3$
M	Megohms	$10^6$
F	Farads	1
UF	Microfarads	$10^{-6}$
PF	Picofarads	$10^{-12}$
H	Henries	1
MH	Millihenries	$10^{-3}$
UH	Microhenries	$10^{-6}$

#### Output Specifications

Once the electrical circuit model of the control system has been described, we need to specify what we want calculated. This specification always starts with a transfer function.

#### Transfer Function

The transfer functions are always specified as a ratio of the voltages across or current through a circuit element at the network output over the input voltage or current. The input voltage or current are specified by the letters VV1 or III1 respectively and either one must appear in the denominator of the transfer function specification. For example, VR3/VV1 used in Fig. 1.2 implies that we want to calculate the ratio of the voltage across R3 to the input voltage V1, i.e., the open circuit transfer voltage function. Since R3 was defined by the statement

R3 4 1 0.8333

node 4 is assumed to be positive with respect to node 1. If R3 has been defined by

R3 1 4 0.8333

the sign of the transfer function would be negative.

When the program is executed using the circuit description shown in Fig. 1.2 and the transfer function specification, the computer generates the output shown in Fig. 1.3. The first information printed under the heading "NUMBER OF LOOPS PER ORDER" describes the complexity of the circuit flowgraph by indicating the number of first order loops, the number of second order loops, etc. This information is only of theoretical interest and can usually be ignored during design. However, since the program prints this data, it will always be included for completeness.

#### Dependent Current and Voltage Sources

NASAP permits use of any of the four possible types of dependent generators (also known as controlled sources):

- a) voltage controlled voltage sources (VCVS)
- b) current controlled current sources (ICIS)
- c) voltage controlled current sources (VCIS)
- d) current controlled voltage sources (ICVS)

Each dependent generator involves two pairs of nodes and two elements: the dependent generator itself and a passive element which defines the controlling voltage or current. An example of each type of dependent source taken from [MO 1] is shown in Fig. 1.4. The node numbers are chosen to indicate that they are elements in a larger circuit and the dependent source is indicated by a diamond to distinguish it from an independent source which will be represented by a circle.

A description of the NASAP coding for dependent generators is in order.

The VCVS shown in Fig. 1.4(a) is specified in NASAP by the instructions found immediately below the figure. The first instruction specifies a dependent voltage source, V3, from node 8 to node 3 (node 3 positive). The value of the source is 867.3 times the voltage across capacitor C2. The

NUMBER OF LOOPS PER ORDER

1= 5  
2= 4

TRANSFER FUNCTION VR3/VV1

$$H(s) = \frac{(5.00E 00 + 1.00E 00 s)}{(3.00E 00 + 3.00E 00 s + 1.00E 00 s^2)}$$

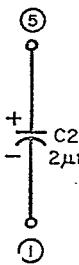
ZEROS OF TRANSFER FUNCTION

ZEROS REAL PART IMAG. PART  
1 -.50000E 01 0.

POLES OF TRANSFER FUNCTION

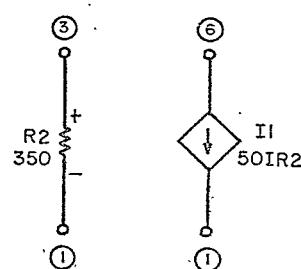
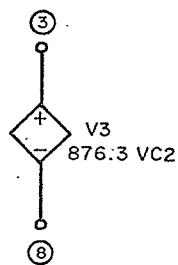
POLES REAL PART IMAG. PART  
1 -.15000E 01 -.86599E 00  
2 -.15000E 01 .86599E 00

Figure 11.3. Transfer Function for RLC Filter



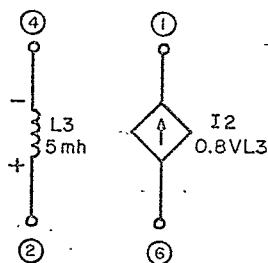
V3 8 3 876.3 VC2  
C2 5 1 2.0UF

(a) VCVS



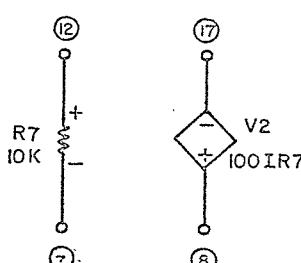
I2 6 1 50 IR2  
R2 3 1 350

(b) ICIS



L3 2 4 5.0MH  
I2 6 1 0.8 VL3

(c) VCIS



R7 12 7 10K  
V2 17 8 100.IR7

(d) ICVS

Figure 1.14. Examples of the Four Types of Dependent Generators and the Corresponding NASAP Instructions

second instruction specifies a capacitor from node 5 to node 1 of  $2.0\mu\text{f}$ . Thus node 5 of the capacitor is assumed positive as noted in Section: Network Element Table, page 1.4. The polarity of the passive controlling element will always be shown on the circuit diagram to help insure that the nodes will be numbered in the correct sequence since the passive element may appear far from the controlled source in the network element table.

Similarly the ICIS shown in Fig. 1.4(b) is specified by the instructions below it. That is, a current source,  $\text{II}_1$ , from node 6 to node 1 with a value of 50 times the current flowing through  $R_2$ .  $R_2$  is defined so that node 3 is assumed positive and thus the current is assumed to flow from node 3 to node 1.

The corresponding NASAP instructions for the VCIS shown in Fig. 1.4(c) and the ICVS shown in Fig. 1.4(d) are written to indicate that it does not matter whether the dependent generator or the passive controlling element is specified first in the network element table.

#### Sensitivity

The NASAP program is capable of determining the sensitivity of the transfer function to changes in the value of any single element. The request for a sensitivity analysis is optional and, if it is desired, is specified following the transfer function specification. This has the typical form:

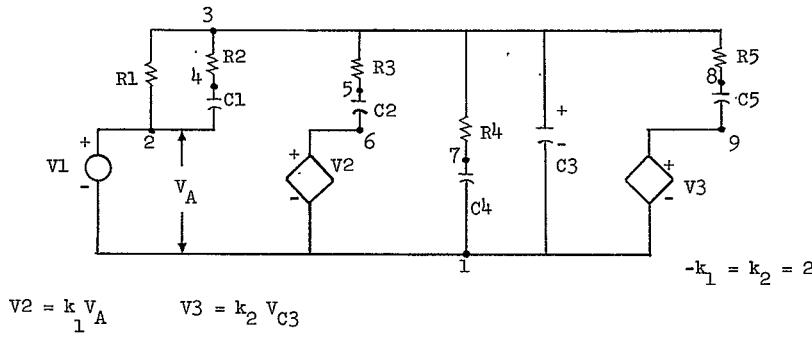
$\text{VR11/VV1/L3}$

This output specification requests the sensitivity of the transfer function  $\text{VR11/VV1}$  with respect to element L3.

The frequency range over which the sensitivity is analyzed is the same as that for which the frequency response is calculated so they are both defined in the same output specification. For the RLC filter of Fig. 1.1, Moe [MO 1] tabulates and displays the response for  $\text{FREQ } -1.2 \ 1.0 \ 0.05$ .

### Further Discussion of Dependent Sources

As stated in Gaertner's Coding manual, dependent sources can be related only to the voltages or currents of passive elements of the flowgraph. However, the controlling current or voltage in the circuit may be that of an active element. Such a situation is illustrated in Fig. 1.5 by the circuit realization for a normalized three-pole low-pass Chebyshev transfer function.



$$V_2 = k_1 V_A \quad V_3 = k_2 V_{C3}$$

Desired Transfer Function:  $V_C3/VV1$

Fig. 1.5

This circuit is an application of a general circuit introduced by Cooper and Harbourt [CO-1]. Note that the controlling voltage,  $V_A$ , of the dependent source  $V_2$ , is the independent voltage source  $V1$ . However if the following card

$V2 \ 1 \ 6 \ -2 \ VV1$

where  $k_1 = -2$

is included in the input list, the error message

INPUT CODING ERROR IN COLUMN 12

will result. The error is the second letter V since the program is only searching for the letter R, L, or C.

However, this difficulty is easily resolved by connecting a resistor R6 across the terminals of V1 (i.e., between nodes 2 and 1) and by making the voltage source dependent on the current through R6. The current through R6 (IR6) equals  $V_1/R_6$ .

Thus  $V_1 = (R_6)IR_6$ . Since  $V_2 = k_1 V_1$ , then

$$V_2 = k_1 \cdot R_6 \cdot IR_6$$

Thus the dependency value between  $V_2$  and  $IR_6$  is  $k_1 \cdot R_6$ . Note that  $R_6$  can have any nonzero numerical value. If, as is the case in this example,  $R_6$  is chosen to have a value of 4 ohms, input list must include the following two cards;

R6 2 1 4  
V2 1 6 -8 IR6

(see input list in Fig. 1.8)

The presence of this extra resistor neither increases the complexity of the flowgraph by generating additional loop sets nor does it affect any of the electrical properties of the original network.

This can be seen by comparison of Figs. 1.6 and 1.7. Fig. 1.6 is a partial flowgraph for the original network in Fig. 1.5. Fig. 1.7 is a partial flowgraph for the addition of  $R_6$  to the network.

Note that in both Figs. 1.6 and 1.7 there is only one path from the voltage node of  $V_1$  to the voltage node of  $V_2$ .

The computer results for the circuit in Fig. 1.5 are given in Figure 1.8.

A similar procedure is followed when a dependent source is a function of the current of a current source. In this case a resistor  $R_s$  is added in series with

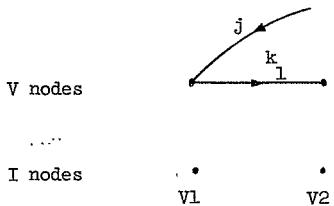


Fig. 1.6.

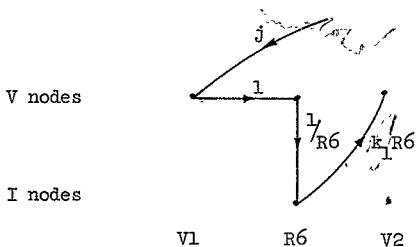


Fig. 1.7

the current source. The dependent source is then made a function of the voltage across this resistor--the dependency value being  $k_1/R_s$  where  $k_1$  is the dependency value between the dependent source and the current source. As in the previous case addition of this resistor to the network under consideration, will in no way affect the flow graph or the electrical properties of the original network.

NASAP PROBLEM COOPER AND HARBOUR CCT

HERTZ  
NONE

V1 1 2 1.  
R1 2 3 4.08  
C1 2 4 .1220F  
R2 4 3 4.08  
V2 1 6 -8 IR6  
C2 6 5 -.2453F  
R3 5 3 4.08  
C3 3 1 1F  
R4 3 7 1.95  
C4 7 1 .5125F  
V3 1 9 2 VC3  
R5 8 3 .307  
C5 9 8 1.63F  
R6 2 1 4

NUMBER OF LOOPS PER ORDER

OUTPUT  
VC3/VV1  
FREQ -2. -0.75 .01  
EXECUTE

1= 13  
2= 35  
3= 37  
4= 16  
5= 2

1/3

TRANSFER FUNCTION VC3/VV1

$$( -4.92E 03 +7.39E 03 S +2.47E 03 S^2 +1.00E 00 S^3 )$$

$$H(S) = 1.984E-04 * ---$$

$$( -9.77E -01 +3.93E 00 S +6.16E 00 S^2 +6.19E 00 S^3 +3.98E 00 S^4 +1.00E 00 S^5 )$$

ZERO OF TRANSFER FUNCTION

POLE OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

POLE REAL PART IMAG. PART

1 -0.1000E 01	0.0000E 00	1 -0.24761E 00	0.96444E 00
2 -0.1998E 01	0.0000E 00	2 -0.24761E 00	-0.96444E 00
3 -0.24621E 04	0.0000E 00	3 -0.99910E 00	0.00000E 00
		4 -0.49402E 00	0.00000E 00
		5 -0.19957E 01	0.00000E 00

Figure 1.8

## IB ERROR MESSAGES

Here are provided a few comments on the meaning of selected error messages available in NASAP.

In subroutine GETCON, there are two error messages.

The message

ALGORITHM FAILURE REVIEW CIRCUIT CODING RULES

usually occurs when the NASAP program is unable to select a legitimate tree from the available resistors or capacitors. This case exists when the candidate type 1 or 2 elements form an R or C loop or when all elements connected to a node are type 6 or 7 elements; i.e., these excluded elements from a cut set. If, for example, the transfer function  $VR_4/VV_1$  is desired for the circuit in Fig. 1.9.

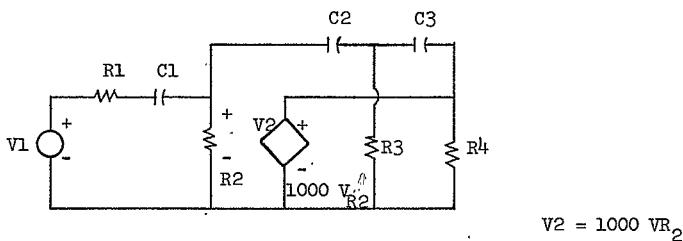


Fig. 1.9

$$V2 = 1000 VR_2$$

then the above error message will be printed out. This is necessitated by the loop formed by the elements  $V_2$  and  $R_4$  which must be in the tree. This difficulty can easily be resolved by calling for the transfer function  $VV_2/VV_1$  which is identical to  $VR_4/VV_1$ .

The other error message is

FLOWGRAPH HAS TOO MANY CONNECTIONS EXECUTION TERMINATED

The connections of the primitive flowgraph are stored in the two dimensional connection matrix LCONC. The size of this matrix is such that it is assumed that no more than nine (edges) branches will emanate from a single flowgraph node. Thus if ten or more branches must emanate from a node, the above error message will be printed and execution stopped. This limitation is completely arbitrary and exists to restrict the size of the matrix LCONC. If one wishes to increase the maximum number of node connections, then the number 10 which appears twice on lines 4010, 5460, and 6220 and which appears once on line 5110 should be changed to a suitable integer N which is defined as the maximum number of node connections (M) plus one. Also the numeral 9 on line 6250 should be changed to the integer M. When these changes are made, the above error message will result when the number of node connections exceeds M instead of when the number of connections exceeds nine.

In subroutine FSORL, there is the error message

FLOWGRAPH FIRST ORDER LOOPS EXCEED 927.

This limitation on the actual loops of the flowgraph, the so-called first order loops, is completely arbitrary. It is necessary to store the first order loops since the higher order loops are determined from these first order loops. However, if this error message is printed out, it usually means that the tree selected was a rather poor choice. A more complete explanation of tree selection to minimize the number of loops of a flowgraph will be given later.

In subroutine BODE the error message is

PROGRAM RESTRICTS GRAPH TO 250 STEPS.

The subscript for the evaluation of the transfer function at various frequencies cannot exceed 250. Thus, if the information on the FREQ input card results in more than 250 frequency evaluations, the NASAP program will not perform any of the calculations included in subroutine BODE including the sensitivity calculations (if this has been called for). The program will jump to the time response calculation if this has been called for. Otherwise, execution will cease after the above error message.

In subroutine SENS there is the error message

SENSITIVITY PLOTS RESTRICTED TO 120 POINTS.

If the input information on the FREQ card results in more than 120 frequency points, the sensitivity results in tabular form will be printed out. However the 3-curve plot and the phase sensitivity plot will be deleted.

At present, extensive diagnostics and debugging capabilities are lacking. Input data can be written on a field-free format, and a circuit tree selection need not be specified by the user since it is done internally by the program. As is sometimes the case, when a proper tree cannot be found, a message is printed out indicating the difficulty and where it occurs in the circuit. Detailed discussion of the tree selection algorithm is found in Chapter II.

Summary

This section details the differences in input data and output results of the University of Pennsylvania version as compared to the standard NASAP 69/I as described in the Gaertner coding manual. [GA-1]

Input Cards

The NASAP version used at the Moore School of Electrical Engineering in the University of Pennsylvania (hereafter called the MSE-NASAP version) requires the user to supply two additional data cards immediately after the first data card (the NASAP PROBLEM card). The first of these cards indicates the frequency units to be used in the evaluation of the transfer function. There are four permissible entries for this card (starting in column 1)

RADIANS

HERTZ

CYCLES PER SECOND

NONE

Only the first two letters (those underlined) must be correct since the NASAP program checks only these letters. Any other information can be included on this card since the program evaluates only the information contained in columns 1 and 2.

The second card indicates the type of time response desired by the user. There are four choices (starting in column 1)

IMPULSE RESPONSE

STEP RESPONSE

RAMP RESPONSE

NONE

The programs again only evaluate the data in columns 1 and 2.

Table 1.2 shows the input listing necessary on the MSE-NASAP version to find the step response for the circuit of Fig. 1.10. The response is the voltage across C1 and the excitation is a step voltage with a magnitude of 3 volts. The BODE plots of the specified transfer function are not being requested.

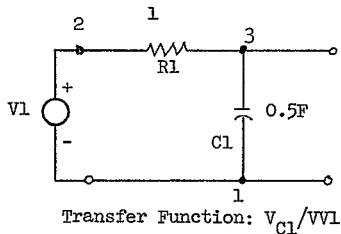


Fig. 1.10

Table 1.2 .

NASAP PROBLEM

NONE

STEP RESPONSE

V1 1 2 3.

R1 2 3 1

C1 3 1 0.5F

OUTPUT

$V_{C1}/VV_1$

TIME 2.0

EXECUTE

### Printed Output Results

The MSE-NASAP version prints out much data on the primitive flowgraph which is used within the computer to determine the user-specified transfer function. The additional output data is printed between the printed listing of the input data cards and the NUMBER OF LOOPS PER ORDER table--both of which are printed by the standard NASAP package.

As an example, Fig. 1.11 gives printed output of a legitimate input listing for circuit 6 introduced in Hutton's report. [HU 1] Fig. 1.12 gives the 5 computer sheets of additional data printed out by the MSE-NASAP version based on the listing of Fig. 1.11. On the sheet immediately following the input list sheet are printed two matrices. The first matrix is the compressed cut set matrix. This matrix has  $(b + 2)$  rows and  $(\lambda + 2)$  columns where

$$b = \text{number of branches in the tree} = n - 1$$

$$n = \text{number of nodes in circuit}$$

$$\lambda = \text{number of links in the co-tree}$$

It must be recalled that

$$b + \lambda = \lambda$$

where  $\lambda$  is the number of elements in the circuit (For the circuit described in Fig. 1.11,  $\lambda = 30$  and  $n = 15$ ). This matrix is useful in that it shows which circuit elements have been selected as tree branches by the NASAP tree-selection algorithm.

As each circuit element is inputted, it is assigned an integer flag (beginning with unity). Thus for the circuit listing of Fig. 1.11.

NASAP PROBLEM 6 FROM HUTTIN REPORT

NONE  
NONE

V1 1 2 1.  
R6 7 1 390  
R15 8 1 25  
R1 2 3 .6.9K  
R2 3 4 6.8K  
R3 4 5 100  
R4 5 6 100K  
R5 5 7 100  
R7 6 1 1.2K.  
R8 9 1 100K  
R9 9 7 100  
R10 10 9 100  
R11 10 12 6.4K  
R12 12 11 6.4K  
R13 13 1 3.4K  
R14 6 8 100  
R16 8 11 100K  
R17 11 1 1K  
C1 3 1 50F  
C2 4 6 .015 5  
C3 10 1 .0050F  
C4 10 12 100F  
C5 13 11 100F  
C6 12 1 200F  
C7 6-11 .0120F  
I1 6 5 100 1212  
I2 1 9 100 1210  
I3 11 8 100 1214  
V2 1 14 1.0 VS7  
V3 1 15 1.0 VS13  
OUTPUT  
VR17/VV1  
EXECUTE

Fig. 1.11

## COMPRESSED CUTSET MATRIX

TOP ROW REPRESENTS CO-TRIE LINKS  
 LEFT COLUMN REPRESENTS TREE BRANCHES  
 INTEGERS REFER TO PLACE IN INPUT LISTING

0	4	5	6	4	11	12	13	14	15	17	22	23	25	26	27	28	0
1	-1	0	0	0	0	0	0	0	0	0	0	0	0	0	0	2	
9	0	-1	0	1	0	0	0	1	0	0	1	0	0	0	2		
15	0	0	0	0	0	0	0	0	-1	1	0	0	0	0	2		
18	0	0	0	0	0	0	0	-1	0	-1	0	-1	0	0	1	2	
29	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	2	
30	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	2	
19	-1	1	0	0	0	0	0	0	0	0	0	0	0	0	0	3	
20	0	-1	1	0	0	0	0	0	0	0	0	0	0	0	0	3	
21	0	0	0	0	1	1	0	0	0	1	0	0	0	0	0	3	
24	0	0	0	0	1	0	-1	1	1	0	0	0	0	0	0	3	
2	0	0	0	-1	0	0	0	0	0	0	0	0	0	0	0	4	
3	0	0	0	0	0	0	0	0	-1	0	0	0	0	0	-1	4	
7	0	0	-1	1	0	0	0	0	0	0	0	0	-1	0	0	4	
10	0	0	0	0	1	-1	0	0	0	0	0	0	-1	0	0	4	
0	4	4	6	4	4	6	4	3	3	3	6	6	6	0			

## FLOWGRAPH CONNECTION MATRIX

1	4	0				
2	-5	-11				
3	-16	17	0			
4	12	0				
5	9	-19	20	0		
6	7	-20	26	0		
7	-6	8	0			
8	2	-7	-2	0		
9	-5	8	16	25	29	0
10	11	-12	0			
11	2	-10	0			
12	10	-21	27	0		
13	-21	24	0			
14	18	-24	0			
15	-22	23	30	0		
16	3	-9	24	0		
17	-3	18	0			
18	1	-14	-17	-23	-25	0
19	-4	5	0			
20	-5	6	1			
21	12	13	22	0		
22	15	-21	0			
23	-15	16	0			
24	-13	14	0			
25	-9	15	0			
26	7	0				
27	10	0				
28	3	-16	0			
29	0					
30	0					

Fig. 1.12

FIRST ORDER LINES	CIRCUMSTANTIAL NUMBER
1	1 4 17 5 9 8 2 -11 -10 -12 -21 13 26 14 18 1
2	1 4 19 6 9 7 4 -11 -10 -12 -21 22 15 23 15 1
3	1 4 19 5 6 10 3 17 13 1
4	1 4 19 5 9 16 28 3 17 18 1
5	1 4 17 5 14 12 -1 -1
6	1 4 15 2 7 18 1
7	1 4 15 2 6 7 2 -11 -10 -12 -21 13 24 14 18 1
8	1 4 15 3 20 7 2 -11 -10 -12 -21 22 15 23 18 1
9	1 4 19 5 21 6 7 8 -9 16 3 17 18 1
10	1 4 19 5 25 6 7 8 -9 16 29 3 17 18 1
11	1 4 19 5 21 6 7 8 -9 16 29 -1 -1
12	1 4 19 5 25 6 7 8 -9 25 18 1
13	1 4 19 5 25 6 7 8 -9 25 18 1
14	1 4 19 5 20 6 7 8 -9 25 18 1
15	1 4 19 5 20 6 7 8 -9 25 18 1
16	1 4 19 7 20 6 25 7 8 -9 16 28 3 17 18 1
17	1 4 19 7 20 6 25 7 8 -9 16 28 3 17 18 1
18	1 4 19 7 20 6 26 7 8 -9 25 18 1
19	2 -2 2
20	2 -6 -7 -6 -23 -5 9 16 3 17 15 -14 -24 -13 -21 12 10 11 2
21	2 -6 -7 -6 -23 -5 9 16 3 17 15 -14 -24 -13 -21 12 10 11 2
22	2 -6 -7 -6 -23 -5 9 16 3 17 15 -14 -24 -13 -21 12 10 11 2
23	2 -6 -7 -6 -23 -5 9 16 3 17 15 -14 -24 -13 -21 12 10 11 2
24	2 -8 -7 -6 -23 -5 9 16 28 3 17 18 -14 -24 -13 -21 12 10 11 2
25	2 -8 -7 -6 -23 -5 9 16 28 3 17 18 -14 -24 -13 -21 12 10 11 2
26	2 -8 -7 -6 -23 -5 9 17 28 3 17 18 -14 -24 -13 -21 12 10 11 2
27	2 -8 -7 -6 -23 -5 9 16 26 3 17 18 -14 -24 -13 -21 12 10 11 2
28	2 -8 -7 -6 -23 -5 9 14 26 7 8 -14 -24 -13 -21 12 10 11 2
29	2 -8 -7 -6 -23 -5 9 14 26 7 8 -14 -24 -13 -21 12 10 11 2
30	2 -8 -7 -6 -23 -5 9 16 28 7 8 -14 -24 -13 -21 12 10 11 2
31	2 -8 -7 -6 -23 -5 9 16 28 7 8 -14 -24 -13 -21 12 10 11 2
32	2 -8 -7 -6 -23 -5 9 25 18 -14 -24 -13 -21 12 10 11 2
33	2 -8 -7 -6 -23 -5 9 25 18 -14 -24 -13 -21 12 27 10 11 2
34	2 -8 -7 -6 -23 -5 9 25 18 -14 -24 -13 -21 12 10 11 2
35	2 -8 -7 -6 -23 -5 9 25 18 -14 -24 -13 -21 12 27 10 11 2
36	2 -8 -7 -6 -23 -5 9 25 18 -14 -24 -13 -21 12 27 10 11 2
37	2 -8 -7 -6 -23 -5 9 25 18 -14 -24 -13 -21 12 27 10 11 2
38	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 10 11 2
39	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 27 10 11 2
40	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 10 11 2
41	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 27 10 11 2
42	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 10 11 2
43	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 27 10 11 2
44	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 10 11 2
45	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 27 10 11 2
46	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 27 10 11 2
47	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 27 10 11 2
48	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 27 10 11 2
49	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 27 10 11 2
50	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 27 10 11 2
51	2 -8 -7 -6 -23 -5 9 17 18 -14 -24 -13 -21 12 27 10 11 2
52	2 -11 2
53	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
54	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
55	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
56	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
57	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
58	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
59	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
60	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
61	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
62	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
63	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
64	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
65	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
66	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
67	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
68	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
69	2 -11 -10 -12 -21 1 3 24 14 13 -17 -3 -16 -9 -5 26 6 7 8 2
70	3 17 16 -25 -16 16 3
71	3 17 16 -25 -16 16 3
72	-4 19 -4
73	-3 18 -5
74	-3 18 -5
75	-3 18 -5
76	-5 20 -2
77	-2 19 6 7 4 -9 -5
78	-2 19 6 20 1 b -9 -5
79	-5 20 6
80	-5 20 6
81	-5 20 6
82	-7 21 7
83	-7 21 7
84	-7 21 7
85	-7 21 7 -18 -25 -9
86	-9 24 -9
87	-10 11 -10
88	10 -12 10
89	10 -12 10
90	12 -21 12
91	12 -21 12
92	-13 -21 22 15 23 18 -14 -24 -13
93	-13 -21 22 15 23 18 -14 -24 -13
94	13 23 14 15 -23 -15 -22 -21 13
95	14 -14 -14
96	14 -14 -14
97	15 -22 15
98	-15 23 -12
99	-17 13 -17
100	1b -23 1b
101	1b -25 1b
102	-21 22 -21

Fig. 1.12b

ORIGIN	CHARGE	LUMINOSITY	LUMINOSITY
2n	33		
3n	143		
4n	347		
5n	500		
6n	349		
7n	124		
8n	11		
2e	33		
3e	143		
4e	346		
5e	500		
6e	347		
7e	127		
8e	12		
2n	31		
3n	176		
4n	500		
5n	521		
6n	341		
7n	232		
8n	15		
2e	2	5	52
3e	50		
4e	237		
5e	734		
6e	1213		
7e	1109		
8e	200		
2n	12		
3n	3		
4n	59		
5n	406		
6n	1355		
7n	4741		
8n	2772		
2e	12		
3e	3		
4e	27		
5e	50		
6e	500		
7e	1674		
8e	3236		
2n	42		
3n	247		
4n	525		
5n	32		
6n	2	5	5
7n	2	13	5
8n	2	13	5
2e	207		
3e	587		
4e	2547		
5e	6137		
6e	4450		
7e	2107		
8e	1557		
2n	563		
3n	124		
4n	125		
5n	507		
6n	2592		
7n	4188		
8n	4338		
2e	3184		
3e	2222		
4e	1434		
5e	5334		
6e	52		
7e	22		
2n	2	42	55
3n	2	47	75
4n	2	47	55
5n	2	454	
6n	2223		
7n	4623		
8n	17524		
2e	3454		
3e	6452		
4e	3415		
5e	2034		
6e	323		
7e	333		
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n	2	25	9
6n	2	25	9
7n	2	25	9
8n	2	25	9
2e	2	25	9
3e	2	25	9
4e	2	25	9
5e	2	25	9
6e	2	25	9
7e	2	25	9
2n	2	25	9
3n	2	25	9
4n	2	25	9
5n</td			

NUMBER OF LOOPS PER ORDER

1= 107

2= 1060

3= 6150

4= 22593

5= 53142

6= .98832

7= 110332

8= .78636

9= 32424

10= 6150

11= 532

415515

TRANSFER FUNCTION 17/VVI

$$( 1.35E 10 \quad +3.05E 14 S \quad +2.62E 14 S \quad 2 \quad -4.67E 12 S \quad -4.25E 06 S \quad +1.00E 00 S \quad 3 \quad 4 \quad 5 )$$

$$H(S) = 8.659E 05 S$$

$$( 7.32E 21 \quad +8.08E 20 S \quad +5.69E 19 S \quad 2 \quad +3.72E 16 S \quad +8.27E 16 S \quad +3.28E 12 S \quad +9.06E 05 S \quad +1.00E 00 S \quad 3 \quad 4 \quad 5 \quad 6 \quad 7 )$$

ZERO OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

1	-0.5648E 02	0.00000E 00
2	-0.109937E 07	-3.62900E 00
3	-0.47039E 01	-3.93309E 00
4	-0.14251E 02	-0.60000E 00
5	0.31554E 07	0.00000E 00

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

<u>Element</u>	<u>Integer Flag</u>
V1	1
R6	2
R15	3
.	.
.	.
.	.
V3	30

An element in the compressed ~~outset~~ matrix can have one of three values -1, 0, or +1. The top and bottom rows and the left-most and right-most columns of the compressed ~~cutset~~ matrix are used for identification purposes described in the following paragraphs.

The non-zero entries in the left-most column are the integer flags of those circuit elements which have been selected as tree branches. The corresponding entry in the right-most column gives the tree hierarchy value of the circuit element (see description of tree-selection algorithm). Hence in Fig. 1.12 we see, that V1(flag = 1) and R7(flag = 9) are tree branches and both have a hierarchy value of 2.

Similarly the non-zero entries in the top row are the integer flags of the circuit elements that are links of the co-tree. The bottom row gives the hierarchy value of these elements. For instance, R1(flag = 4) and R2(flag = 5) are links of the co-tree and have hierarchy values of 4.

The information contained in the compressed ~~cutset~~ matrix is used to develop the flowgraph connection matrix which is a mathematical description of the primitive flowgraph. This matrix consists of  $\lambda$  (= number of elements) rows. The entries in the left-most column represent either the voltage node or the current node of the circuit elements (depending upon whether the elements are branches or links respectively) identified by the integer flags numbered consecutively from 1 to  $\lambda$ .

If the left-most entry of a particular row of the flowgraph connection matrix represents a voltage node of an element, then each of the other entries of this row indicates a connection from the voltage node of the left-most entry to the voltage node of the corresponding entry. There is one exception--if one of these entries refers to a voltage controlled current source, then there is a connection from the voltage node of the left-most entry to the current node of this particular entry. Conversely, if the left-most entry of a particular row of the flowgraph connection matrix represents a current node of an element, then each of the other entries of this row indicates a connection from the current node of the left-most entry to the current node of the corresponding entry. There is one exception--if one of these entries refers to a current controlled voltage source, then there is a connection from the current node of the left-most entry to the voltage node of this particular entry. A few examples from Figure 1.12 will clarify this description. The second row of the flowgraph connection matrix in Figure 1.12 contains

2      -8      -11      0

From the cutset matrix it is seen that the element with integer flag 2 (i.e. R6) is a tree branch. Thus there are two connections (with a value -1) from the voltage node of R6 - one to the voltage node of R5(flag = 8) and the other to the voltage node of R9(flag = 11). The zero indicates the end of the row.

As a second example, the fifth row of this connection matrix contains

5      9      -19      20      0

Since R2(flag = 5) is a link in the co-tree, there is a connection with a value of +1 from current node of R2 to the current node of R7(flag = 9). Also there is a connection (value = -1) to the current node of C1(flag = 19) and one (with value = -1) to the current node of C2(flag = 20). Use of Ohm's Law yields the passive element (R,L,C) joining the voltage and current nodes, since the voltage across an element is related to the current flowing through the element by means

of the element's impedance.

The next printed output consists of all the closed loops (first order loops) contained in the flowgraph determined by means of the flowgraph connection matrix. The heading of this output block of data is

#### FIRST ORDER LOOPS BY CONSECUTIVE LOOPS

Each loop is defined by the nodes contained in the loop. This output shows the order in which the nodes have been found by NASAP for each loop. Each loop is given an identification integer starting with unity. This identification integer is shown in the left-most columns. The remaining integers in a given row refer to the integer flags of the circuit elements.

In Figure 1.12 we see that the flowgraph under discussion has a total of 102 first order loops. Let us examine one of these loops more closely--say loop 81 defined by

-6      26      7      -6

The first and last integers are always identical since it is the starting point of the loop. From the input listing we have

Element	Integer Flag
R3	6
I1	26
R4	7

Each loop is found by the path-finding procedure on the flowgraph connection matrix (for details see the Potash-McNamee User's Manual). From the cutset matrix we note that R4 is a tree branch while I1 and R3 are co-tree links.

To illustrate this we now show how this loop 81 can be found from the flowgraph connection matrix. Start at the current node of R3 (R3 is a link). There is a connection from this node to the current node of the current source I1 (i.e., an integer 26 in row 6 of the connection matrix). There is a connection from the current node of I1 to the current node of R4 (i.e., an integer 7 in the 26th

row of the connection matrix). Since R4 is a passive element, there is a connection (with a value = R4) from the current node of R4 to the voltage node of R4 (recall that R4 is a tree branch). From the 7th row of the connection matrix (integer flag of R4 = 7) we observe that -6 is an entry. Thus there is a connection from the voltage node of R4 to the voltage node of R3 (with a value of -1). Since R3 is a passive element that is a co-tree link, there is a connection (with value =  $1/R_3$ ) from the voltage node of R3 to the current node of R3 (the starting point)--thus completing the loop.

The higher order loops, the sets of non-touching first order loops, are easily found by assigning each first order loop an integer value based upon the nodes contained in the loop (each node is identified by the circuit element integer flag). This integer value is stored in the one-dimensional array LOOP defined by

$$\text{LOOP}(J) = \sum_{k=1}^n 2^{|N_k| - 1}$$

where J is the loop identification integer

$N_k$  refers to the kth flowgraph node in the Jth loop

n = number of flowgraph nodes in the Jth loop

As an example, for the loop numbered 79 in Fig. 1.12, we have

$$\text{LOOP}(79) = 2^{-6} - 1 + 2^7 - 1 = 2^5 + 2^6$$

$$\text{LOOP}(79) = (96)_{10}$$

$$= (110000)_2$$

(Note: see Potash-McNamee Manual for details of how higher-order loop are found by use of the array LOOP).

Finally the sheet with the heading

ORDER OF LOOP

LOOP NUMBERS

is printed out to assist the user in locating the point of termination if the number of flowgraph loops is so large that the allowable computer time is used up before completion of the analysis.

The information of this sheet is presented in this manner:

1. Every 50th second order loop is printed out with the identification integers of the first order loop comprising the second order loop.
2. When the number of some  $j$ th order loop ( $j = 2, 3, 4, \dots$ ) equals 500, the number of all loops of all order at this point of the loop enumeration procedure. (Note: see the Potash-McNamee Manual for details of the HIGORL subroutine which determines the higher order loops from the first order loops).

From Figure 1.12 we see that the output data

2        18        88

is the fourth of this type in the output data. Consequently the second order loop consisting of loop 18 and loop 88 is the 200th second order loop found by the NASAP subroutine called HIGORL. (Note: from the list of first order loops, loop 88 consists of nodes 10 and 12 which do not appear as nodes in loop 18). Immediately following this line of output, we observe that the number of 9th order loops found by NASAP equals 500. When this occurred, NASAP had found 209 2nd order loops, 987 3rd order loops, etc.

It should be noted that, if a flowgraph has less than 50 second order loops and the number of each of the  $n$ th ( $n = 3, 4, 5, \dots$ ) order loops of this flowgraph is less than 500, only the heading

ORDER OF LOOP                          LOOP NUMBERS

will be printed in the MSE-NASAP version.

In order to find all the higher order loops and at the same time to avoid repeating any of these loops, the HIGORL subroutine selects each first order

loop in order of its identification integer and determines all the higher order loops formed by this selected first order loop and those loops having identification integers greater than that of the selected loop (see Potash-McNamee manual for details).

With this procedure in mind, we can use the higher-order loop data printed out by the MSE-NASAP version to determine approximately where the HIGORL subroutine was in the higher order loop finding process if execution is terminated before the desired transfer function is obtained. This will enable the user to determine whether the problem should be executed with a longer running time specified or should be cancelled since the generated flowgraph is too complicated (with regard to running time) for the computer in use.

As an example, let us assume that the printed output of Figure 1.12 ended with the line

2        68        88

There was no sheet with the heading

NUMBER OF LOOPS PER ORDER

and no sheet listing the specified transfer function. From the list of first order loops, we see that this flowgraph contains 102 loops. From the above output line, we see that the loop under selection by HIGORL is the loop numbered 68. In other words, when this line was printed loops 69 through 101 have not yet been selected as the starting loop in the higher order loop finding process.

Since the number of higher order loops generated by the starting loop decreases as the identification integer of the starting loop increases, knowledge that loop 68 (in a group of 102 loops) is the present starting loop enables the user to decide whether or not to rerun the problem. It must be noted that the last line printed does not mean that this was the last loop set found by the HIGORL subroutine before termination. It is possible that as many as 49 more

2nd order loops and many more nth ( $n = 3, 4, 5 \dots$ ) order loops have been found since the printing of the last output line and before the termination.

Following this output data, the output results of the MSE-NASAP version conform with the output data of NASAP 69/I. One remaining minor difference is the order of the sensitivity output data.

The order for NASAP 69/I is:

Table of Sensitivities of  $\text{Re}H$ ,  $\text{Im}H$ ,  $|H|$ , Phase H as a function of frequency.

2. Table of the Logarithm of the above sensitivities as a function of frequency.
3. Plot of 3 Sensitivity Expressions.
4. Plot of the Logarithm of the Sensitivity.
5. Table of the Sensitivity Function as a function of frequency.
6. Plot of the Logarithm of the Absolute Value of the Sensitivity function.
7. Plot of the Phase of the Sensitivity Function.
8. Pole and Zero Sensitivities.

On the other hand, the order of the MSE-NASAP version is:

1. Table of Sensitivity Function as a function of frequency.
2. Plot of Logarithm of Absolute Value of Sensitivity Function.
3. Plot of Phase of Sensitivity Function.
4. Table of Sensitivities of  $\text{Re}H$ ,  $\text{Im}H$ ,  $|H|$ , Phase H as a function of frequency.
5. Table of Logarithms of the above sensitivities as a function of frequency
6. Plot of 3 Sensitivity Expressions.
7. Plot of the Logarithm of the Sensitivity of the Phase H.
8. Pole and Zero Sensitivities.

## CHAPTER II

### NASAP TREE SELECTION ALGORITHM--USER OPTIONS

#### IIA GENERAL DESCRIPTION

Although the graph representation of an electric network usually has a large number of possible trees (i.e., a structure containing  $n-1$  branches which interconnect the  $n$  nodes of the circuit without forming any closed paths), NASAP has an algorithm that selects only a particular tree configuration. This tree is the basis for subsequent circuit analysis.

Each electrical circuit element is assigned a Type number as shown.

Independent voltage source	1
Dependent voltage source	2
Capacitor	3
Resistor	4
Inductor	5
-----	
Dependent current source	6
Independent current source	7

Elements of type 1 and 2 are always included in the tree while type 6 and 7 elements are never included in branches of a tree. If there are not enough elements of type 1 and 2 to form a tree, a search is made of type 3 elements (i.e. capacitors) starting with the first capacitance listed in the input data and working down the input list. If a tree is still not found after searching through all the type 3 elements, a similar search is made of all type 4 elements (selecting those type 4 elements that do not form closed paths and neglecting those that do). If a tree does not result, a search is made of type 5 elements. If a tree is not found after this search, an error message will be printed.

The element type categories 2 and 6 need a further explanation. The elements in category 2 include not only dependent voltage sources, but also those elements

whose voltages control the voltage or current of some dependent source or whose voltage is the required output variable or input variable. For example, the following is a legal NASAP input record.

I3 3 4 5.2 VR3

Since the voltage across resistor R3 controls the dependent current source I3, resistor R3 will be assigned element type 2 not element type 4. Also the following is a legal NASAP output record

VL5/VVL/R2

Since the voltage across inductor L5 is the desired output variable, Element type 2 not element type 5 will be assigned to inductor L5. Similarly since V1 is the input, it will be assigned element type 2 not element type 1.

Similarly the elements in category 6 include not only dependent current sources but also those elements whose currents control the voltage or current of some dependent source or whose current is the required output variable or input variable.

In

I3 3 4 5.2 IR3

Resistor R3 is assigned element type 6 since the current through R3 controls the dependent current source I3. In

IC5/IIZ

capacitor C5 is assigned element type 6 since the current through C5 is the required output current. Also, I2 will be a type 6 element (not type 7) since it is the input variable.

Due to the search technique that reads down the input list looking for elements to be branches of a tree, the actual tree selected by NASAP can be varied simply by rearranging the order in which the elements are placed in the input list (NOTE: There is no restriction as to the order in which the elements are listed in NASAP). However, not all of the possible tree configurations can be selected by NASAP due to the requirement that all type 1 and 2 elements must be included in the tree while all type 6 and 7 elements cannot be included in the tree.

## IIB ILLUSTRATIVE EXAMPLE

The complexity of the flowgraph is greatly dependent on the particular tree. By complexity is meant the number of loop sets present in the flowgraph. The values of these loop sets are used in NASAP to calculate the transfer function by means of Mason's formula as extended by Happ for the closed signal-flow graph. Since much time is consumed by the NASAP algorithms in finding the loop sets, selection of the tree that minimizes the number of loop set can save computer time as well as increase the accuracy of the coefficients of the polynomials in the transfer function.

In Fig. 2.1 is shown a transistor with known h-parameters. The input resistance of the circuit is to be calculated.

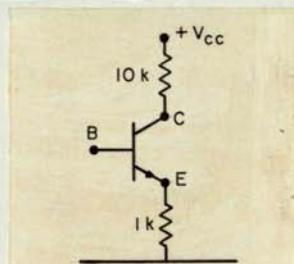


Fig. 2.1. A Common Emitter Transistor

The NASAP equivalent circuit model is given in Fig. 2.2.

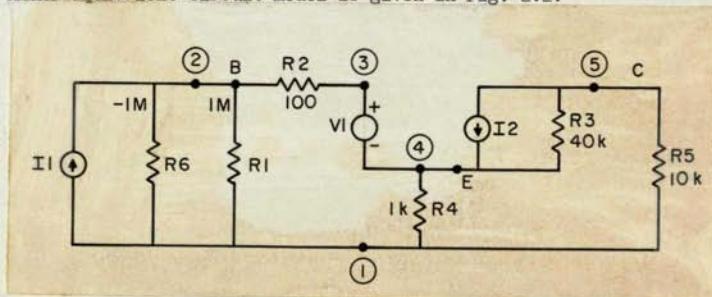


Fig. 2.2. Equivalent Circuit for Transistor in Fig. 2.1.

$I_1$  is the independent current source.  $R_1$  and  $R_6$  are positive and negative resistances of equal numerical value. These resistances, when added to the network, yield an element whose voltage equals the input voltage and at the same time does not load the circuit (the parallel connection of a positive and negative resistance yields an infinite resistance). Thus the required transfer function for the input impedance is

$$VR_1/I_1$$

The network in Fig. 2.2 has five nodes; thus the tree for this circuit must have four elements. The elements  $R_1$ ,  $V_1$ , and  $R_3$  are type 2 elements and must be included in the tree while  $I_1$ ,  $I_2$ , and  $R_2$  are type 6 elements and cannot be part of the tree. Thus one more element must be selected to form the tree. The partially completed tree structure is shown in Fig. 2.3.

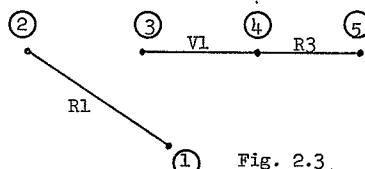


Fig. 2.3.

$R_6$  cannot be used as a tree element since it forms a closed path with  $R_1$  which is an element in tree. However either  $R_5$  or  $R_4$  can be used to form a tree. Thus the NASAP tree selection algorithm will pick that resistance which is listed first in the input data. Thus if the input data is listed as follows,

```

I1 1 2 1.
I2 5 4 100 IR2
V1 4 3 0.0001 VR3
R1 2 1 1M
R2 2 3 1K
R3 5 4 40K
R5 5 1 10k
R4 4 1 1K
R6 2 1 -1M
OUTPUT
VR1/I1

```

NONE

NONE

C1 1 2 1,

I2 5 4 1CO IR2

V1 4 3 0.0001 VR3

R1 2 1 1M

R2 2 3 1K

R3 5 4 40K

R4 4 1 1K

R6 2 1 -1M

R5 5 1 10K

OUTPUT

VR1/III

EXECUTE

NUMBER OF LOOPS PER ORDER

1= 9

2= 16

3= 4

TRANSFER FUNCTION VR1/III

$$( \quad 1.00E \ 00 \quad )$$

$$H(S) = 8.033E \ 04 * \frac{1}{( \quad 1.00E \ 00 \quad )}$$

ZERO OF TRANSFER FUNCTION

NONE

POLE OF TRANSFER FUNCTION

NONE

Fig. 2.7

NASAP IIT PROBLEM

with  $\pm 1M$  resistors

NONE  
NONE

I1 1 2 1.  
I2 5 4 100 IR2  
V1 4 3 0.0001 VR3  
R1 2 1 1M  
R2 2 3 1K  
R3 5 4 40K  
R5 5 1 10K  
R4 4 1 1K  
R6 2 1 -1M  
OUTPUT  
VR1/III1  
EXECUTE

NUMBER OF LOOPS PER ORDER

1= 14  
2= 29  
3= 10

TRANSFER FUNCTION VR1/III1

(  
1.00E 00 )

H(S)= 8.033E 04\*-----

(  
1.00E 00 )

ZERO OF TRANSFER FUNCTION

NONE

POLE OF TRANSFER FUNCTION

NONE

Fig. 2.5

R5 will be included in the tree elements. The chosen tree is shown in Fig. 2.4.

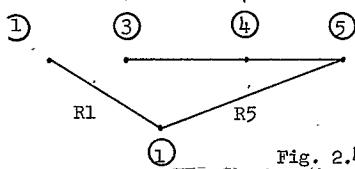


Fig. 2.4

The output results are given in Fig. 2.5.

However, if the input listing format is given as

```
11 1 2 1.  
I2 5 4 100 IR2  
V1 4 3 0.0001 VR3  
R1 2 1 IM  
R2 2 3 IK  
R3 5 4 40k  
R4 4 1 1K  
R6 2 1 -1M  
R5 5 1 10K  
OUTPUT  
VR1/IR1
```

(Note that R5 is now  
the last element de-  
scribed)

The tree selected by NASAP will now contain R4 as shown in Fig. 2.6.

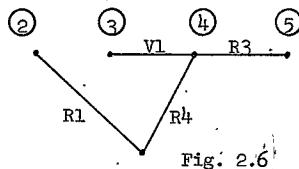


Fig. 2.6

The output results are shown in Fig. 2.7.

Note that both coding formats yield the same transfer function, as expected.

However, the tree with R4 yields a flowgraph with 29 loop sets as compared with 53 loop sets in the flowgraph obtained from the tree with R5. Thus the number of loop sets is almost halved by the proper choice of a tree for this circuit

configuration. This reduction of loop sets is apparent in the execution time. The tree with R5, NASAP required an execution time of 24.3 seconds on the RCA Spectra 70/46 while the execution time for the tree with R4 was 22.6 seconds, a saving of 1.7 seconds.

## TIC THE OPTIMUM TREE

There is a procedure to select the optimum tree (optimum in the sense that such a tree will minimize the total number of loops in the primitive flowgraph) that does not require a great deal of effort on the part of the NASAP user.

The details and proof of this procedure are given in reference [Z0-1]. The procedure begins by putting all type 1 and 2 elements in the tree. Next all capacitors (type 3 elements) are included in the tree. If a capacitor forms a loop with some type 1 or 2 elements, it is removed from the tree. If two capacitors form a loop with some other tree branches, then the user arbitrarily picks one of these capacitors to be a tree branch--realizing that the capacitor picked to be the tree branch must precede the other capacitor in the input lists. All type 6 and 7 elements and any element that forms a loop with the chosen tree branches are put in the co-tree. Note that if the sum of the type 1, 2, and 3 elements equals the number of nodes minus one and if these elements do not form any loops, then the NASAP tree selection algorithm will have picked a tree after a search of all capacitors. The user thus will not be able to vary the tree.

If, on the other hand, the number of resistors in the circuit is greater than the number of elements necessary to complete the tree, the NASAP user will have some flexibility in the type of tree selected by NASAP by permitting the resistor input cards.

Once the user has selected a tree, the optimum tree search procedure goes as follows:

Each link forms a loop with some of the branches of the tree. For each link, the number of tree branches in each loop is recorded as well as the specific branches that form the loop. The Branch Count is the sum of the number of tree branches in each loop. The Circuit Count is the sum of the number of loops for a specified tree branch. The Branch Count will always equal the Circuit Count. However the tree that yields a smaller Branch Count will yield a flowgraph with fewer loops.

As an example let us reconsider the input impedance circuit given above.

Recall that it was shown that  $V_1$ ,  $R_1$ , and  $R_3$  must be included in the tree while  $I_1$ ,  $I_2$ ,  $R_2$  and  $R_6$  must be links in the co-tree. However either  $R_4$  or  $R_5$  can be in the tree depending on the order of the input cards. If  $R_4$  is a branch of the tree, then there are five links in the co-tree -  $I_1$ ,  $I_2$ ,  $R_2$ ,  $R_6$ , and  $R_5$ .

The link  $I_1$  forms a loop with branch  $R_1$

The link  $I_2$  forms a loop with branch  $R_3$

The link  $R_2$  forms a loop with branches  $R_1$ ,  $V_1$ ,  $R_4$

The link  $R_6$  forms a loop with branch  $R_1$

Finally, the link  $R_5$  forms a loop with branches  $R_3$ ,  $R_4$ . Table 2.1 summarizes this information. ("Branch" refers to a tree branch.)

Table 2.1

Links	$I_1$	$I_2$	$R_2$	$R_6$	$R_5$	branch count
Branches/Loop	1	1	3	1	2	8
Branches		$V_1$	$R_1$	$R_3$	$R_4$	circuit count
Loops/Branch		1	3	2	2	8

Note that the Branch Count equals the Circuit Count, as required.

However, if  $R_5$  is made a tree branch, then the co-tree links are  $I_1$ ,  $I_2$ ,  $R_2$ ,  $R_6$ ,  $R_4$ .

The link  $I_1$  forms a loop with branch  $R_1$

The link  $I_2$  forms a loop with branch  $R_3$

The link  $R_2$  forms a loop with branches  $R_1$ ,  $V_1$ ,  $R_3$ , and  $R_5$

The link  $R_6$  forms a loop with branch  $R_1$

The link  $R_4$  forms a loop with branches  $R_3$  and  $R_5$

Table 2.2 summarizes this information.

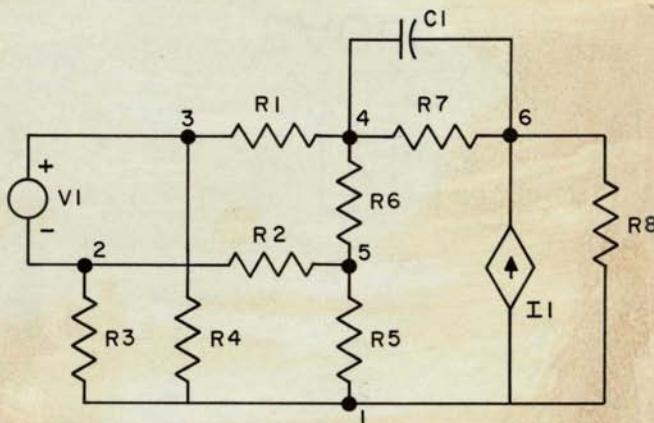
Table 2.2

Links	I1	I2	R2	R6	R4	Branch count
Branches/Loop	1	1	1	4	2	9
Branches		V1	R1	R3	R5	Circuit count
Loops/Branch			1	3	3	9

Both the Branch Count and Circuit Count equal nine. Since the Branch Count for the tree containing  $R_4$  is less than that for the tree with  $R_5$ , then the number of loops for the flowgraph formed from the tree with  $R_4$  will be less than that obtained from the tree with  $R_5$ . As noted above, there are 29 loop sets including 9 first order loops in the flowgraph formed from the  $R_4$  tree as opposed to the 53 loop sets including 14 first order loops in the flowgraph formed from the  $R_5$  tree.

#### Illustrative Example

The circuit shown in Fig. 2.8 from [MO-1] illustrates the case when different trees yield the same branch count.



$$I_1 = 500 V_{R6} / V_1$$

Desired transfer function  $VR_8/VV_1$

Fig. 2.8. An EOG Filter from [MO-1]

It is apparent that V1, R6, and R8 (type 2 elements) and C1 must be branches of the tree while I1, R5, and R7 must be links in the co-tree since they form loop with the above mentioned tree branches. For the six node circuit in Fig. 2.8 a tree has five branches. Since four of these branches have already been designated the last branch can be chosen from resistor R1, R2, R3 or R4. Thus four different trees can be formed by the NASAP algorithm. The question remains which of these four trees will yield the fewest loops in the flowgraph.

If R2 is selected as the fifth tree branch, then the links of the co-tree are I1, R5, R7, R1, R3, and R4. Table 2.3 gives the pertinent information on the number of branches in each loop formed by each link and the number of loops that each branch is a part of.

Table 2.3: Tree with R2

Links	I1	R5	R7	R1	R3	R4
Branches/Loop	1	3	1	3	4	5
Branches	V1	R6	R8	C1	R2	
Loops/Branch	2	4	4	4	3	

The Branch Count for the R2 tree is 17.

If instead we let R4 be a branch of the tree; then the links become I1, R5, R7, R1, R2, and R3. Table 2.4 gives the corresponding branch and loop information.

Table 2.4: Tree with R4

Links	I1	R5	R7	R1	R2	R3
Branches/Loop	1	3	1	3	5	2
Branches	V1	R6	R8	C1	R4	
Loops/Branch	2	2	4	4	3	

Here the Branch Count is 15. Since the branch count for the R4 tree is smaller, then this tree is a better choice than the R2 tree in terms of fewer loops for the flowgraph.

If we next let R3 be a tree branch, the co-tree links become I1, R5, R7, R1, R2, and R4. Table 2.5 gives the branch and loop data for this case.

Table 2.5: Tree with R3

Link	I1	R5	R7	R1	R2	R4
Branches/Loop	1	3	1	4	4	2
Branch	V1	R6	R8	C1	R3	
Loop/Branch	2	2	4	4	3	

Note that the branch count is 15, the same as that obtained for the R4 tree.

Finally Table 2.6 gives the branch and loop data for the case when the tree contains the element R1.

Table 2.6: Tree with R1

Link	I1	R5	R7	R2	R3	R4
Branches/Link	1	3	1	3	4	3
Branch	V1	R6	R8	C1	R1	
Loops/Branch	2	2	4	4	3	

The Branch Count for the R1 tree is also 15.

Thus three of the four trees have the same low Branch count. Thus the other criteria, the Branch product and the Loop Product, must be used to determine which of these three trees will yield the flowgraph with the fewest loops. The Loop Product, defined as the product of the number of loops involving each branch, is the same for the three trees; namely 192. However, the Branch Product, defined as the product of the number of tree branches in each loop formed by each co-tree link, is different in each case. The Branch Product is  $1 \times 3 \times 1 \times 3 \times 5 \times 2 = 90$  for the R4 tree while the Branch Product for the R3 tree is  $1 \times 3 \times 1 \times 4 \times 4 \times 2 = 96$ . The Branch Product for the R1 tree is  $1 \times 3 \times 1 \times 3 \times 4 \times 3 = 108$ . It is found that the tree having the smallest Branch Product (R4) yields the flowgraph with the fewest

Note that the Branch Counts and Branch Products of both trees are equal. Since the Loop Product of the C2-C5 tree is smaller, it would seem that this tree would yield the flowgraph with the fewer loops.

However, the modified Branch count of the C3-C6 tree is smaller than that of the C2-C5 tree. This criterion indicates that the C3-C6 tree gives the fewer flowgraph loops. This, in fact, is the case as the computer results of Figs. 2.18, 2.19, 2.20, 2.21 show. The tree with C3 and C6 generates a flowgraph of 969 loops including 19 first order loops while there are 1529 loops including 25 first order loops in the flowgraph formed from the C2-C5 tree.

From Durin and Chan, [DU 1] it has been shown that the star tree (a tree in which all the branches have a common node) yields the minimum number of flowgraph loops. The more star-like the tree structure, the fewer the number of loops in the corresponding flowgraph.

In the circuit of Fig. 2.17, all of the type 2 elements are connected together at node 1. Thus 5 of the 9 branches are joined at a single node. Examination of this circuit reveals that two other elements (R4 and R8) are also connected to node 1. If these resistors were branches of the tree, then 7 of the 9 tree branches would be connected to a single node--a tree structure that is definitely more star-like than the two trees described above. However, since R4 and R8 are resistors and type 4 elements, they are not considered for eligibility as tree branches by the NASAP tree selection algorithm until all type 3 elements (i.e. capacitors) are considered. As is shown above, the capacitors are so connected in the circuit that they do form legitimate trees (in fact, 4 trees depending upon the input listing). Thus R4 and R8, as type 4 elements, can never be tree branches.. We next indicate how to overcome this problem.

If somehow the voltages across R4 and R8 controlled some dependent sources, they would become type 2 elements and would therefore be branches of the

NASAP FROM 1-STAGE FNG FILTER MOF

NONE  
NONE

V1 2 3 1.0  
R4 3 1 1K  
R2 2 5 10K  
R3 2 1 1K  
R1 3 4 10K  
R5 5 1 500K  
R6 5 4 1M  
R7 4 6 500K  
R8 6 1 200  
C1 4 6 0.015UF  
I1 1 6 500 VR6  
OUTPUT

VR8/VVI  
EXECUTE

NUMBER OF LOOPS PER ORDER

1= 72  
2= 151  
3= 95  
4= 14

332

TRANSFER FUNCTION VR8/VVI

(  
( 1.24E 04 +1.00E 00 S )

H(S)=-5.382E-01\*-----

(  
( 1.33E 02 +1.00E 00 S )

ZERO OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

1 -0.12388E 05 0.00000E 00

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

1 -0.13346E 03 0.00000E 00

Autosimotic  
(1)

Figs. 2.9 and 2.10: R4 in tree

NASAP FROM 1-STAGE FOG FILTER MOE

NONE  
NONE

V1 2 3 1.0  
R3 2 1 1K

R2 2 5 10K

R1 3 4 10K

R4 3 1 1K

R5 5 1 500K

R6 5 4 1M

R7 4 6 500K

R8 6 1 200

C1 4 6 0.015UF

I1 1 6 500 VR6

OUTPUT

VR8/VV1

EXECUTE

NUMBER OF LOOPS PER ORDER

1= 74

2= 151

3= 93

4= 16

334

TRANSFER FUNCTION VR8/VV1

(  
( 1.24E 04 +1.00E 00 S )

H(S)=-5.382E-01\*

(  
( 1.33E 02 +1.00E 00 S )

ZERO OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

1 -0.12388E 05 0.00000E 00

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

1 -0.13346E 03 0.00000E 00

Figs. 2.11 and 2.12: R3 in tree

NASAP FROM 1-STAGE EOG FILTER MOE

NONE

NONE

V1 2 3 1.0

R1 3 4 10K

R2 2 5 10K

R3 2 1 1K

R4 3 1 1K

R5 5 1 500K

R6 5 4 1M

R7 4 6 500K

R8 6 1 200

C1 4 6 0.015uF

I1 1 6 500 VR6

OUTPUT

VR8/VV1

EXECUTE

NUMBER OF LOOPS PER ORDER

1= 80

2= 163

3= 104

4= 19

366

TRANSFER FUNCTION VR8/VV1

$$( \quad 1.24E 04 \quad +1.00E 00 S \quad )$$

$$H(S) = -5.382E-01j$$

$$( \quad 1.33E 02 \quad +1.00E 00 S \quad )$$

ZERO OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

$$1 -0.12388E 05 \quad 0.00000E 00$$

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

$$1 -0.13345E 03 \quad 0.00000E 00$$

Figs. 2.13 and 2.14: R1 in tree

NASAP FROM 1-STAGE EDG FILTER MOE

NONE

NONE

V1 2 3 1.0

R2 2 5 1.0K

R1 3 4 1.0K

R3 2 1 1K

R4 3 1 1K

R5 5 1 50.0K

R6 5 4 1M

R7 4 6 500.0K

R8 6 1 200

C1 4 6 0.015UF

I1 1 6 500 VR6

OUTPUT

VR8/VV1

EXECUTE

NUMBER OF LOOPS PER ORDER

1= 186

2= 369

3= 210

4= 31

796

TRANSFER FUNCTION VR8/VV1

$$\left( \begin{array}{c} \\ \left( \begin{array}{cc} 1.245 \text{ E } 04 & +1.000 \text{ E } 00 \text{ S } \end{array} \right) \end{array} \right)$$

H(S)=-5.382E-01\*

$$\left( \begin{array}{c} \\ \left( \begin{array}{cc} 1.238 \text{ E } 02 & +1.000 \text{ E } 00 \text{ S } \end{array} \right) \end{array} \right)$$

ZERO OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

1 -0.123E7E 05 0.00000E 00

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

1 -0.13346E 03 0.00000E 00

Figs. 2.15 and 2.16: R2 in tree

loops. Conversely, the tree having the largest Branch Product (R1) yields the flowgraph with the greatest number of loops when compared with the flowgraphs generated by the R4 and R3 trees. However, the flowgraph generated by the R1 tree will give fewer loops than the flowgraph generated by the R2 tree since the R1 tree yields a smaller branch count.

The circuit in Table 2.1 was analyzed using the NASAP program for the four possible trees available from the NASAP tree selection algorithm. The computer results are given in Figs. 2.9-2.16. Fig. 2.9 gives an input listing that includes R4 in the tree. Note that this is not the only listing order that will cause R1 to be a branch of the tree--the only requirement on the input listing is that the card describing R4 must precede those describing R1, R2, and R3. Note that there are 332 flowgraph loops including 72 first order loops. Fig. 2.12 shows an input listing that makes R3 a tree branch and show that there are 334 flowgraph loops including 74 first order loops. In Fig. 2.14, is a listing with R1 in the tree. The flowgraph loops generated by the R1 tree total 366 including 80 first order loops. Finally Fig. 2.16 gives a listing with R2 a tree branch. The flowgraph loops number 796 including 186 first order loops. Figs. 2.10, 2.11, 2.14, 2.15 give the transfer functions obtained for the different trees..

Note that the R4 tree does indeed yield the fewest flowgraph loops. However there is almost no difference between the number of loops generated by the R3 tree and the R4 tree. The R1 tree yields about 10% more loops than either the R4 tree or the R3 tree. The R2 tree generates more than twice as many loops as either of the other trees.

#### A Needed Modification

It should be noted that the primitive flowgraph as developed by the NASAP program does differ slightly from the actual primitive flowgraph developed from Kirchhoff's voltage and current laws and Ohm's Law. In the NASAP flow-

graph there can be no connections to the current node of an independent or dependent voltage source as well as no connections to the voltage node of an independent or dependent current source. However such connections may exist in a true primitive flowgraph. If there are no connections emanating from these nodes, then these nodes will not be a part of any loop and no information will be lost in determining the transfer function. This is the reason why one is not able to call for the current of a voltage source or the voltage of a current source as an output variable in the NASAP program. Furthermore, this difference between the NASAP and true primitive flowgraph affects the procedure for determining the optimum tree.

Since there can be no connection between the voltage and current nodes of current and voltage sources (either dependent or independent), let us modify the branch count by omitting the branch count of those loops which are formed from independent and dependent current sources and by deleting from the branch count of each remaining loop those branches representing independent or dependent voltage sources. Similarly the Loop Count will be modified by omitting the loop count of those tree branches which represent independent or dependent voltage sources and by deleting those loops, which are formed by independent and dependent current sources from the loop count of the remaining tree branches. In other words, the modified branch count is the sum of the passive element tree branches that are part of those loops formed from the passive element links while the modified loop count is the sum of the loops, formed from passive element links, that pass through passive element tree branches. Note that the modified loop count will always equal the modified branch count.

The need for the modified Branch Count, Branch Product, and Loop Product is demonstrated from the analysis of the Butterworth filter circuit in Fig.  
2.17. [SA-1]

Desired Transfer Function:  $VV_3/VV_1$

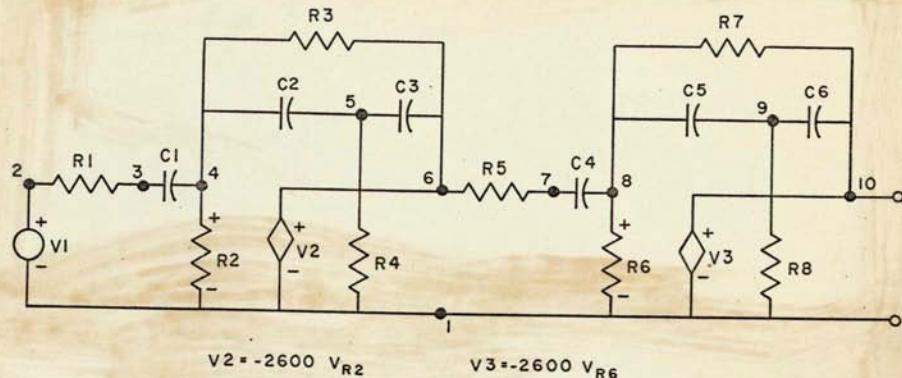


Fig. 2.17: Equivalent Circuit of a Butterworth Filter

The type 2 elements,  $V_1-R_2-V_2-R_6-V_3$  will be tree branches as will  $C_1$  and  $C_4$  due to the topology of the network. Either  $C_2$  or  $C_3$  but not both can be branches and the same situation holds for  $C_5$  and  $C_6$ . There are four trees to choose from, namely

1. with  $C_2$  and  $C_5$
2. with  $C_3$  and  $C_6$
3. with  $C_2$  and  $C_6$
4. with  $C_3$  and  $C_5$

We will concern ourselves only with the first two choices. Table 2.7 gives branch-loop data for the  $C_3-C_6$  tree while the same data for the  $C_2-C_5$  tree is shown in Table 2.8.

Table 2.7: Tree with C3 and C6

Link		R1	R3	R4	C2	R5	R7	R8	C5
		3	2	2	3	3	2	2	3

Branch	V1	V2	V3	C1	R2	C3	C4	R6	C6
	1	4	3	1	3	2	1	3	2

Branch Count = 20  
 Branch Product = 1296  
 Loop Product = 432

Passive Link	R1	R3	R4	C2	R5	R7	R8	C5
	2	1	1	2	2	1	1	2

Passive Branch	C1	R2	C3	C4	R6	C6
	1	3	2	1	3	2

Modified Branch Count = 12  
 Modified Branch Product = 16  
 Modified Loop Product = 36

Table 2.8: Tree with C2 and C5

Link		R1	R3	R4	C3	R5	R7	R8	C6
		3	2	2	3	3	2	2	3

Branch	V1	V2	V3	C1	R2	C2	C4	R6	C5
	1	3	2	1	4	2	1	4	2

Branch Count = 20  
 Branch Product = 1296  
 Loop Product = 384

Passive Link	R1	R3	R4	C3	R5	R7	R8	C6
	2	1	2	2	2	1	2	2

Passive Branch	C1	R2	C2	C4	R6	C5
	1	4	2	1	4	2

Modified Branch Count = 14  
 Modified Branch Product = 64  
 Modified Loop Product = 64

## NASAP FRENCH #1 BUTTERWORTH

NONE  
NONEV1-1 2 -1  
R1 2 3 7.414K

C1-3 -4 -202UF

R2 4 1 10M

C2-4 -5 -20UF

R3 4 -6 10.510K

R4 5 1 .0523K

V2 1 -6 -2600-VR2

R5 6 7 8.542K

C4-7 -8 -282UF

R6 8 1 10M

C5 8 9 -20UF

NUMBER-OF-LOOPS-PER-ORDER

1= 25

2= 186

3= 474

4= 529

5= 266

6= 49

R7-8-10-12-108K

R8 9 1 .06025K

V3-1-10 -2600-VR6

OUTPUT

VV3/VV1

EXECUTE

1529

58.

TRANSFER-FUNCTION VV3/VV1

$$( 0.00E 00 \quad +0.00E 00 \quad S \quad +1.98E 05 \quad S \quad +8.93E 02 \quad S \quad +1.00E 00 \quad S \quad )$$

H(S)= 1.578E 02\*

$$( 4.32E-12 \quad +3.71E-10 \quad S \quad +2.33E-09 \quad S \quad +1.39E-07 \quad S \quad +3.05E-05 \quad S \quad +1.10E-03 \quad S \quad +1.00E-00 \quad S \quad )$$

## ZERO OF TRANSFER FUNCTION

## POLE OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

POLE REAL PART IMAG. PART

1 0.00000E 00 0.00000E 00

1 -0.42975E 01 0.58384E 02

2 0.00000E 00 -0.00000E 00

2 -0.42975E 01 -0.58384E 02

3 -0.41494E 03 0.97890E-09

3 -0.49350E 01 0.67260E 02

4 -0.47801E-03 -0.79611E-08

4 -0.49350E 01 -0.67260E 02

5 -0.41509E 03 -0.17377E-05

6 -0.66777E 03 -0.34203E-06

Figs. 2.18 and 2.19: Tree with C2 and C5

NASA\_FRENCH\_LUITTERULTH

NOINE  
NONE

V1 1 2 1  
R1 2 3 7 .414K

C1 3 4 .202JF

R2 4 1 .101L

C3 5 6 .200UF

C2 4 5 .200UF

R3 4 6 12.516K

R4 5 1 .0522K

V2 1 6 -2600 VR2

R5 6 7 .8 .542K

C4 7 8 .282UF

R6 8 1 .17M

C5 9 10 .200UF

C5 8 9 .200UF

R7 8 10 12.106K

R8 9 1 .04025K

V3 1 10 -2600 VR6

OUTPUT

VV3/VV1

EXECUTE

NUMBER OF LOOPS PER ORDER

969

TRANSFER FUNCTION VV3/VV1

$$( -0.00E 00 \quad +0.00E 00 \quad S \quad +1.90E 05 \quad S^2 \quad +8.93E 02 \quad S^3 \quad +1.00E 00 \quad S^4 )$$

$$H(S) = 1.578E 02 *$$

$$( -4.32E 12 \quad +3.71E 10 \quad S \quad +2.33E 09 \quad S^2 \quad +1.39E 07 \quad S^3 \quad +3.05E 05 \quad S^4 \quad +1.10E 03 \quad S^5 \quad +1.00E 00 \quad S^6 )$$

ZERO OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

1 -0.00000E 00 0.00000E 00

2 0.0000E 00 0.0000E 00

3 -0.41429E 03 0.97490E+09

4 -0.47891E 03 -0.79611E-08

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

1 -0.42956E 01 0.58382E 02

2 -0.42956E 01 -0.58382E 02

3 -0.49374E 01 0.67261E 02

4 -0.49374E 01 -0.67261E 02

5 -0.41512E 03 0.10178E-04

6 -0.41512E 03 0.10862E-04 ← pole 6 equals pole 5

ratio of pole 6 to pole 5 is -6.6777 E-03

Figs. 2.20 and 2.21: Tree with C3 and C6

tree. The problem remains of selecting these dependent sources such that they do not affect the circuit being analyzed. This is easily accomplished by making these sources dependent voltage sources with one node of the source connected to any node of the original circuit and with the other node left unconnected. Each unconnected node is actually an additional node in the modified circuit. Since a tree is an interconnection of all nodes and since the only connection to these hanging nodes is the source itself, then each of these sources must be a tree branch and thus a voltage source. However since one node of these voltage source is left floating, these sources will in no way affect the original circuit because no current can flow through these sources. The zero-load feature of this type of "dummy" source is also quite apparent in the flowgraph. There is a connection to the voltage node of the voltage source from the voltage node of the controlling element. However no connection emanates from the voltage node of the voltage source since this "dummy" voltage source is not involved in any of the network loops formed by the links of the co-tree. A node can be part of a flowgraph loop only if there is a connection leading to and away from the node. Since these "dummy" voltage sources are not part of any flowgraph loop, then they cannot affect the determination of the transfer function.

Thus the addition of the following two cards

V4	1	11	1.0	VR4
V5	1	12	1.0	VR8

to the input lists of either Fig. 2.18 or Fig. 2.20 will make R4 and R8 tree branches without affecting the electrical properties of the original circuit. Nodes 11 and 12 are the floating nodes. Note that the dependency value (1.0 in this case) is completely arbitrary and can have any value except 0.0. In essence, the above "trick" simply is an artificial means to tell the NASAP program what elements we wish to have in the tree. It does have the drawback in that elements are needlessly wasted but this only becomes a factor when the

number of elements in the circuit is near the limit set by the length of the computer word. Note also that if an element which one desires to put into the tree by the above "trick" is also declared a type 6 element somewhere else in the input list (i.e., its current controls some source or is the desired output variable), then an error message will result.

Fig. 2.22 gives the necessary branch data for the R4-R8 tree. Take note that the branch count, branch product, and loop product were determined without regard to the "dummy" sources V4 and V5.

Fig. 2.22: Tree with R4 and R8

Branch count	= 18	Modified Branch count	= 12
Branch product	= 576	Modified Branch product	= 16
Loop product	= 216	Modified Loop product	= 36

Note that the modified data is identical to that obtained for the C3-C6 tree. However the unmodified branch count is two less than that of either the C2-C5 tree or the C3-C6 tree. Figs. 2.24 and 2.25 give the computer results. The R4-R8 tree generates a flowgraph of 737 loops (including 17 first order loops). This is less than one-half the loops generated by the C2-C5 tree and more than 200 loops less than the number of loops derived from the C3-C6 tree--a substantial reduction.

NASAP FRENCH 1 BUFFER PTH

NONE  
NONE

V1 1 2 1  
R1 2 3 7.414E  
C1 3 4 .2024E  
R2 4 1 10M  
C3 5 6 .200E  
C2 4 5 200UF  
R3 4 6 1.0E  
R4 5 1 .0527E  
V2 1 0 -260E VR2  
R5 6 7 6.542E  
C4 7 8 -.2824E  
R6 8 1 10M  
C6 9 1 2.00E  
C5 8 9 200UF  
R7 8 10 -12.100E  
R8 9 1 .06027E  
V3 1 11 -260E V4 1  
V4 1 11 1.0 VR4  
V5 1 12 1.0 VRS

NUMBER GE LIPS BEB IRDEK

737

OUTPUT  
VV3/VV4  
EXECUTE

92

TRANSFER FU C101. VV3/VV1

$$( 0.00E 00 +0.00E 00 S +1.90E 05 S ^2 +8.93E 02 S ^3 +1.00E 00 S ^4 )$$

$$H(S) = 1.578E 029 -$$

$$( 4.32E 12 +3.71E 10 S +2.33E 09 S ^2 +1.35E 07 S ^3 +2.05E 05 S ^4 +1.10E 03 S ^5 +1.00E 00 S ^6 )$$

ROLE OF TRANSFER EQUATION

ZERO OF TRANSFER FU C101.

ROLE REAL PART IMAG. PART

ZERO REAL PART (MAG. PART)

1	-4.0000E 03	+1.0000E 00	1	-0.42914E 01	0.58033E 02
2	3.0000E 03	-6.0000E 00	2	-0.42914E 01	-0.58033E 02
3	-1.0000E 13	-1.0000E 00	3	-0.41513E 03	0.0000E 00
4	-0.47801E 03	0.0000E 00	4	-0.40423E 01	-0.7259E 02

Figs. 2.24 and 2.25

CHAPTER III  
MODELING A CONTROL SYSTEM FOR NASAP

III.A GENERAL DISCUSSION OF CONTROL SYSTEMS

The preliminary discussion of feedback control systems is kept brief on the assumption that the reader already is generally acquainted with feedback control theory. Familiarity with introductory textbooks such as Dorf's Modern Control Systems [DO-1] and Perkins and Cruz Engineering of Dynamic Systems [PE-1] would be particularly helpful in that they use aerospace feedback control system problems as illustrative examples.

The major subdivisions of feedback control systems usually are:

1. A plant, process or controlled system wherein the position or state is being regulated or monitored.
2. The controller consisting of a sensor and control elements.
3. A comparator or error-sensing device to detect the difference between the input reference and the output signal.

Next we itemize the major steps involved in the design of a feedback control system. These steps are:

- a. Establish performance specifications for the system  
(e.g. type of control, tolerance on accuracy, speed of response, overshoot, etc.)
- b. Interpret the specification data in terms of design parameters and components of the control system. (give due consideration to reliability, space, cost, etc.)
- c. Formulation of the transfer functions of the components and analysis of the preliminary design.
- d. Improve the performance of the preliminary design by suitable compensation to meet the specifications.

The scope of computer-aided design in this manual only covers aspects of steps c and d.

Control system design can be carried out either in the frequency or in the time domain. This is important in considering the possible role of NASAP in such applications. Control engineers have found it convenient, in the analysis of linear feedback control systems, to use the transfer function concept and the block diagram representation of the system. The transfer function concept is basic in the application of the frequency response method of analysis. In this approach the steady state response of the system to a sinusoidal input is used. The output/input signal relationship for each component of a control system is described by a transfer function. The operations of these components are represented by noninteracting blocks which are interconnected to form the block diagram or the corresponding signal flow graph of the overall system. Thus one obtains a functional representation of the feedback control system equivalent to the set of simultaneous differential equations that relate the variables of the physical system.

The basic procedures usually followed to analyze and design a feedback control system by the frequency response method are:

- i) Determine transfer functions for each of the components used in the system (from the differential equations via transforms or from physical measurements).
- ii) Formulate the signal flow graph from the system block diagram.
- iii) Reduce the complicated block diagram of the system to a simple single loop configuration having a transfer function for the forward and the feedback branch if the open loop transfer function and output transform of the control system are to be used.
- iv) Determine the system characteristics using the Bode plot or Nyquist diagram (an alternative graphical method uses the Nichols chart).
- v) To have the system meet the prescribed performance specifications, design the necessary compensators that will reshape the plots obtained in step iv. This may involve cut and try.

In addition to the above, the designer often has to investigate sensitivity of parameters to variation of individual elements. The final step may include analog simulation or physical model tests.

As an alternative to the frequency response approach, the analysis and synthesis of feedback control systems determination of the system stability and the evaluation of the output of the system in response to impulse, step, or ramp input functions. Here again there is often the need for compensation of the system so that it will meet specs.

In the remaining chapters of this manual we shall indicate how NASAP can assist the design engineer to accomplish some of this work. It should be noted that some of these procedures are best carried out with a hybrid computer. Dr. C. H. Beck [BE-1] has developed a hybrid NASAP module for such applications as part of this cooperative development of the NASAP program.

IIIIB1 Equivalent Electrical Networks for Transfer Functions

*Copys + l. c.*

Before a control system can be analyzed using NASAP and before any necessary compensation can be determined, the dynamic characteristics of the plant must first be simulated by an electric network which has an equivalent dynamic characteristic. The transfer function of a lumped linear plant can be expressed as a ratio of two polynomials. The problem of modeling the plant transfer function can be simplified if the polynomials are put in factored form. The individual factors or group of factors can be modeled by using simple RLC circuits. Then for a complicated transfer function these circuits are connected in cascade with suitable isolation between each circuit to prevent loading that would result in a change in the modeled transfer function. This necessary isolation is obtained by using ideal dependent voltage or current sources (which are available in NASAP). Table 3.1 gives a list of some elementary circuits with isolation and their corresponding transfer functions.

Table 3.1

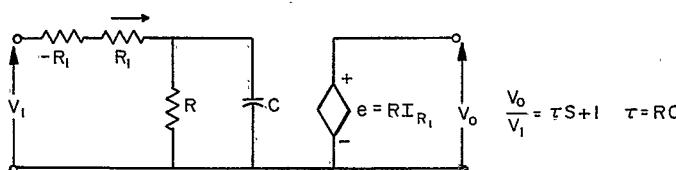
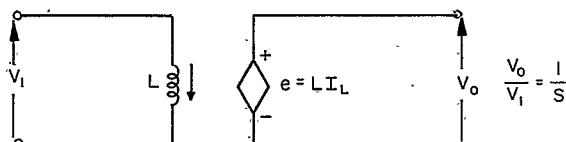
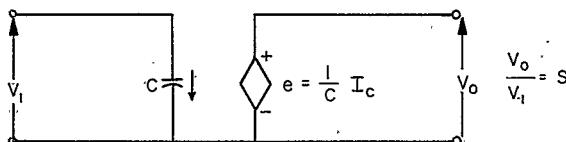
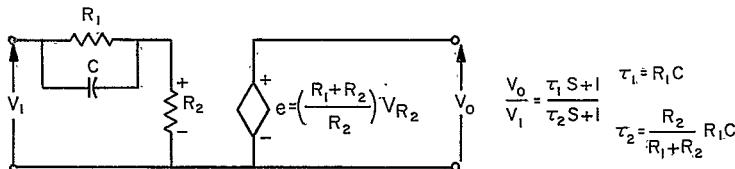
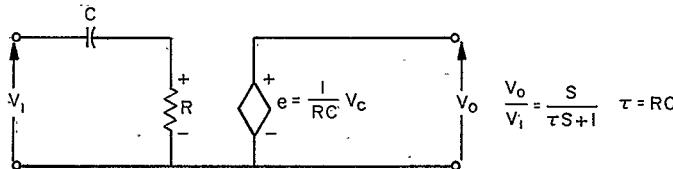
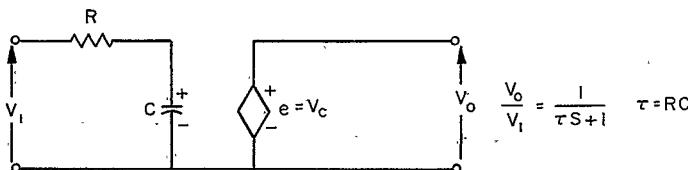
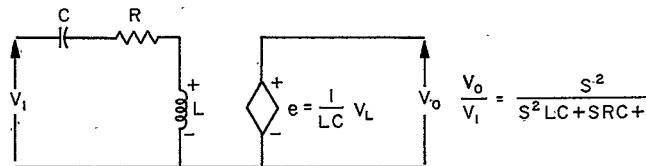
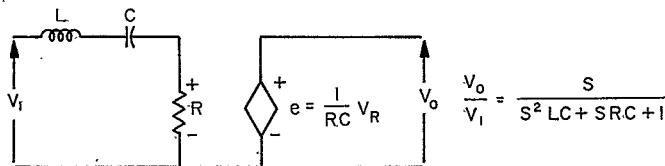
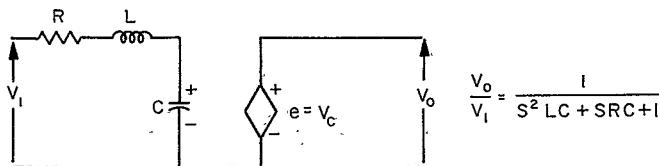


Table 3.1 (continued)



## IIIB2 Cascade Interconnection of Transfer Function Models

As indicated earlier the simple transfer function models can be cascaded.

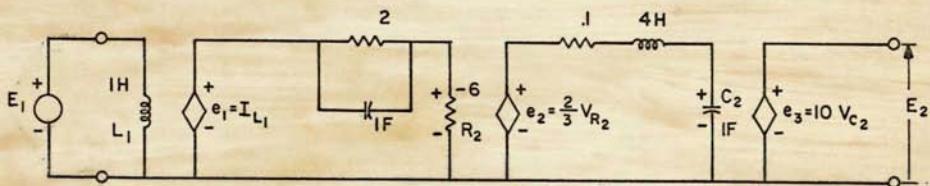
Consider a plant transfer function given as:

$$\frac{E_2(s)}{E_1(s)} = \frac{10(2s + 1)}{s(3s + 1)(4s^2 + 0.1s + 1)} \quad (3.1)$$

This can be rewritten as:

$$\frac{E_2(s)}{E_1(s)} = \frac{1}{s} \cdot \left( \frac{2s + 1}{3s + 1} \right) \cdot \left( \frac{10}{4s^2 + 0.1s + 1} \right) \quad (3.2)$$

Thus a cascade interconnection of three circuits from Table 3.1 can be used to model the above transfer function.



Note that  $R_2$  is a negative resistance and that the gain factor (for this example, 10) is included in the dependency relation for the third dependent voltage source ( $e_3$ ). The gain factor could just as easily be included in the dependency relations of  $e_1$  or  $e_2$ .

Many of the subsequent examples used to illustrate various aspects of computer-aided control system design will incorporate techniques for modeling the pertinent transfer function.

## IIIC ADDITIONAL EQUIVALENT NETWORK MODELS

## IIIC1 Use of Negative R, L or C

Since negative element values of R, L, and C are permitted in the NASAP input coding, rational transfer functions of control systems can be modeled simply by application of the continued fraction expansion procedure. Accordingly the rational function will be represented in general by the input admittance or impedance of a ladder structure consisting of positive or negative R, L, and C elements. It is emphasized that this approach works because physical realizability as a passive network is not a consideration. Only the equivalent dynamic characteristic matters.

As an example, let us consider the biquadratic all-pass function

$$F(s) = \frac{s^2 - as + b}{s^2 + as + b} \quad (3.3)$$

Performing the continued fraction expansion of  $F(s)$  yields

$$\begin{aligned} & s^2 + as + b \quad \overline{\Bigg|} \quad \frac{1}{s^2 - as + b} \\ & \overline{\Bigg|} \quad \frac{s^2 + as + b - \frac{1}{2a}s}{s^2 + as + b} \\ & \overline{\Bigg|} \quad \frac{-2as}{s^2 + as + b} \\ & \overline{\Bigg|} \quad \frac{s^2}{as + b} \quad \overline{\Bigg|} \quad \frac{-2}{-2as - 2b} \quad \frac{\frac{a}{2b}s}{as + b} \\ & \overline{\Bigg|} \quad \frac{2b}{as + b} \\ & \overline{\Bigg|} \quad \frac{as}{b} \quad \overline{\Bigg|} \quad \frac{2}{2b} \end{aligned} \quad (3.4)$$

If  $F(s)$  is assumed to be an admittance  $Y_1$ , then the resulting ladder network is shown in Fig. 3.1.

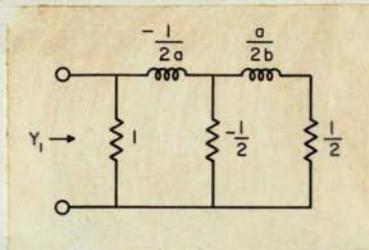
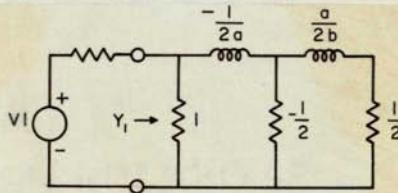


Fig. 3.1

In other words, the input admittance  $Y_1$  of the circuit of Fig. 3.1 is a representation of the specified rational function  $F(s)$ .

To utilize this input admittance model with the NASAP program, two additional elements, a voltage source and a resistance of very small value, must be included in the circuit of Fig. 3.1 (see Fig. 3.2).



$$Y_1 = IRI/VV1$$

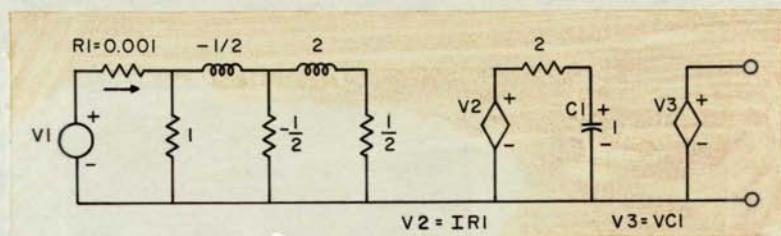
Fig. 3.2

The voltage source  $V1$  represents the excitation for the input admittance  $Y_1 = \frac{I_1}{V_1}$ . The response is the current  $I_1$ . However the current flowing through a voltage source cannot be specified with the NASAP program. Since no element of the circuit of Fig. 3.1 is in series with the source  $V1$ , it is necessary to include the small resistor  $R1$ . Thus the input admittance of this circuit can be specified as

$$Y_1 = IRI/VV1 \quad (3.5)$$

The circuit of Fig. 3.2 can be easily cascaded with other isolated circuits to model more complex rational functions. As an example, the transfer voltage ratio of the circuit of Fig. 3.3 models the function.

$$\frac{s^2 - s + 4}{(s^2 + s + 4)(2s + 1)} \quad (3.6)$$



$$VV3/VV1 = \frac{s^2 - s + 4}{(s^2 + s + 4)(2s + 1)}$$

Fig. 3.3

Alternatively the function  $F(s)$  given above can be assumed to be an input impedance  $Z_1$ . From the continued fraction expansion, we obtain the NASAP applicable circuit of Fig. 3.4.

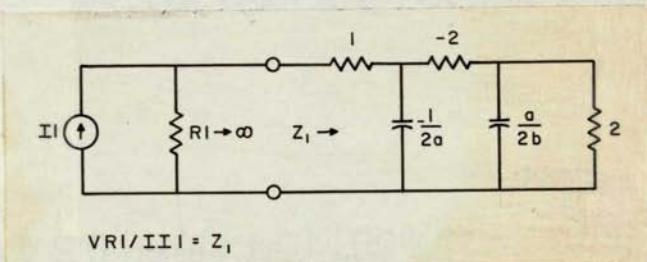


Fig. 3.4

The current source  $I1$  represents the excitation necessary for the input impedance. The response is the voltage across the very large resistance  $R1$  which must be added to the circuit since no single element obtained from the continued fraction expansion is connected across the ideal current source terminals.

However it is possible to avoid the use of the small series resistor of Fig. 3.2 and the large shunt resistor of Fig. 3.4 by taking the reciprocal of (i.e. inverting) the rational function that is to be modelled by NASAP before performing the continued fraction expansion.

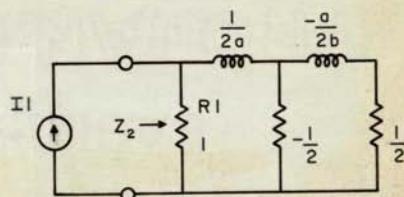
As an example, let us desire to model the biquadratic all-pass function given above by the input impedance of some ladder circuit. Thus we desire a circuit whose input impedance is defined by

$$Z_2(s) = \frac{s^2 - as + b}{s^2 + as + b} \quad (3.7)$$

Inverting this expression we obtain

$$Y_2(s) = \frac{s^2 + as + b}{s^2 - as + b} \quad (3.8)$$

Performing the continued fraction expansion this time yields the NASAP circuit is shown in Fig. 3.5.



$$Y_2 = VRI / IIII = \frac{s^2 - as + b}{s^2 + as + b}$$

Fig. 3.5

Note that now there is no need to include a large shunt resistor across the  $I_{II}$  current source terminals (as in Fig. 3.4) since the resistor  $R_1$  already shunts these terminals.

Similarly, we can model the biquadratic all-pass function by the input

admittance of a ladder network defined by

$$Y_3 = \frac{s^2 - as + b}{s^2 + as + b} \quad (3.9)$$

Inverting this expression and then performing the continued fraction expansion (given above), we obtain the ladder circuit of Fig. 3.6.

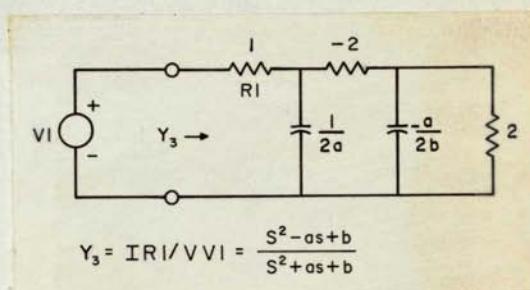


Fig. 3.6

Note that addition of a small series resistor (as in Fig. 3.2) is not needed since the current flowing in the resistor  $R_1$  is from the voltage source  $V_1$ .

In summary, by use of the continued fraction expansion procedure, we have obtained four NASAP-codable circuits to model the specified all-pass function  $F(s)$ . The circuits of Figs. 3.5 and 3.6 are more desirable as models than those of Figs. 3.2 and 3.4 since they require one less element. It should be further noted that each of these four circuits has another desirable feature. The NASAP tree-selection algorithm will automatically select the shunt elements of each circuit as tree branches. This will lead to a star tree which generates the flow-graph with the fewest loops.

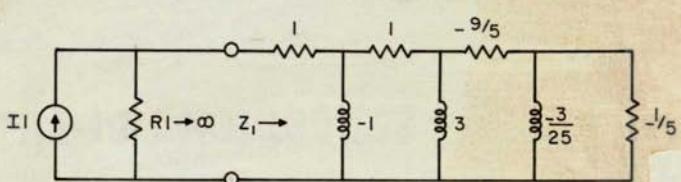
This will always occur if the polynomials of the specified rational function are arranged in descending powers of  $s$ . The resulting continued fraction expansion will either make the series elements inductors or the shunt elements capacitors.

## IIIIC2 Illustrative Examples

Sometimes it is necessary to arrange the polynomials in ascending powers to attain a ladder network with both positive and negative elements. For example, let us assume we wish to model the function

$$F_1(s) = \frac{1}{s^3 + 2s^2 + s + 1} \quad (3.10)$$

as the input impedance of a ladder network. We cannot perform a continued fraction expansion on  $F_1^{-1}(s) = s^3 + 2s^2 + s + 1$ . Consequently we will need a large shunt resistor in the NASAP model. Furthermore we must rearrange the denominator of  $F_1(s)$  in ascending powers of  $s$  before performing the continued fraction expansion. The resulting NASAP model with shunting resistor  $R_1$  is shown in Fig. 3.7. Note that, since the shunt elements are



$$Z_1 = VRI / I1 = \frac{1}{s^3 + 2s^2 + s + 1}$$

Fig. 3.7

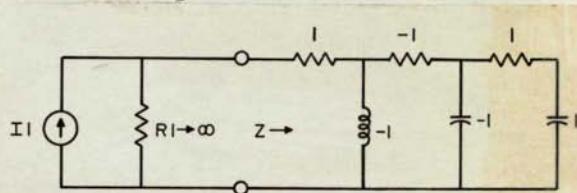
inductors, the NASAP tree-selection algorithm will pick a linear tree consisting of resistor  $R_1$  and three of the other four resistances (the particular  $R$ 's depending upon their location in the input list). This type of tree yields the largest number of loops in the corresponding primitive flowgraph.

Sometimes in the continued fraction expansion procedure more than one term is eliminated by subtraction. This may necessitate a rearranging of the remaining polynomials to achieve a NASAP model. Such a case exists for the function

$$F_2(s) = \frac{1}{s^3 + s^2 + s + 1} \quad (3.11)$$

The first three steps of the continued fraction expansion process is based on ascending order of the denominator. Then three terms become zero. This necessitates reversing the polynomial  $-s^3 - s^2 - s$  to  $-s^3 - s^2 - s$ .

In Fig. 3.8 is given a NASAP-codable ladder network whose input impedance equals the desired function  $F_2(s)$ .



$$Z = VR/I = \frac{1}{s^3 + s^2 + s + 1}$$

Fig. 3.8

Another example is the biquadratic function

$$F_3(s) = \frac{s^2 + as + b}{s^2 + as + c} \quad (3.12)$$

Let us model this function as an input impedance. Inverting  $F_3(s)$  and then performing the continued fraction expansion yields

$$\begin{aligned} & \frac{1}{s^2 + as + b} = \frac{1}{s^2 + as + c} \\ & \frac{s^2 + as + b}{s^2 + as + c} = \frac{b}{c - b} \\ & \left( c - b \right) \frac{s^2 + as + b}{b} = \frac{b}{c - b} \\ & \frac{b}{as + s^2} = \frac{b}{c - b} \\ & \frac{b}{as + s^2} = \frac{c - b}{as} \\ & \frac{c - b}{as + s^2} = \frac{c - b}{c - b} \\ & \frac{c - b}{as + s^2} = \frac{-a^2}{c - b} \end{aligned} \quad (3.13)$$

$$\frac{as}{s^2} \left( -\frac{c-b}{a} \right) = \underline{\underline{-\frac{c-b}{a}s}}$$

(3.13)  
cont.

The resulting network is shown in Fig. 3.9.

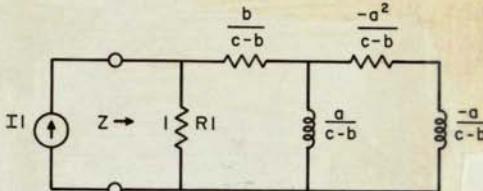


Fig. 3.9

$$Z = VRI / IIII = \frac{s^2 + as + b}{s^2 + as + c}$$

Note the large shunt resistor is again avoided by use of the initial inversion. Note also that in the underlined section of the continued fraction process it was necessary to reverse the polynomical  $s^2 + as + b$  since two terms were eliminated in the preceding subtraction.

### IIIC3 Equivalent Networks for Summing Element

Since most control systems require some sort of feedback loop, an electrical network equivalent to the summing (or subtracting) element that is compatible with NASAP must be used. Such networks are shown in Figs. 3.10a and 3.10b.

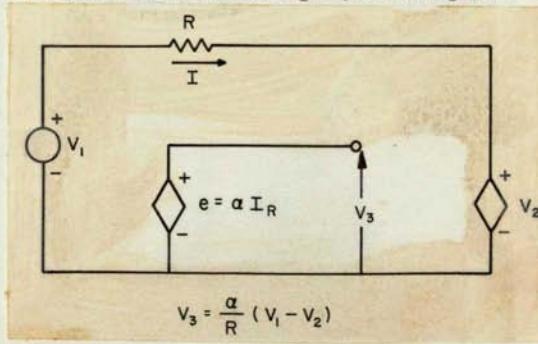


Fig. 3.10a

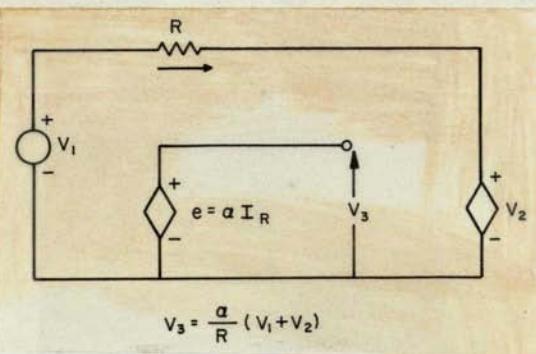


Fig. 3.10b

In Fig. 3.10a if  $\alpha = R = 1$ , the output voltage is equal to the difference of the two input voltage. Similarly if  $\alpha = R = 1$  in Fig. 3.10b, the output voltage is equal to the sum of the two input voltages.

## IIID MODELS OF FEEDBACK CONTROL SYSTEMS

## IIIDL Examples of System Models

Thus we now have all the elements necessary to model a feedback control system with an electric network that is compatible with NASAP. As an example, the unity feedback control system shown in Fig. 3.11.

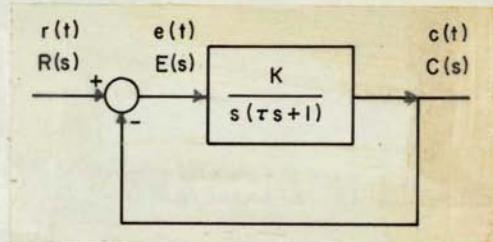


Fig. 3.11

has the following equivalent electric network

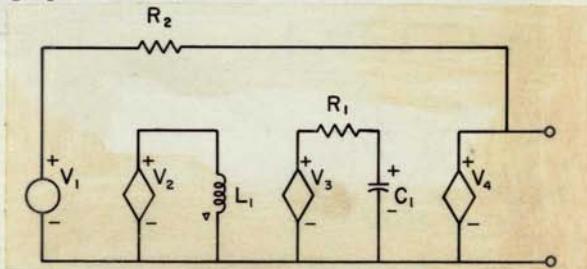


Fig. 3.12

where  $V_1$  is equivalent to  $r(t)$

$V_2$  is equivalent to  $e(t)$

$V_4$  is equivalent to  $c(t)$

A second example shows a NASAP model for a non-unity feedback system (Fig. 3.13).

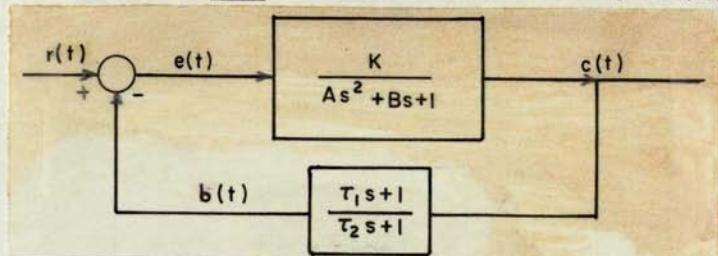


Fig. 3.13

Fig. 3.14 is the equivalent electric network.

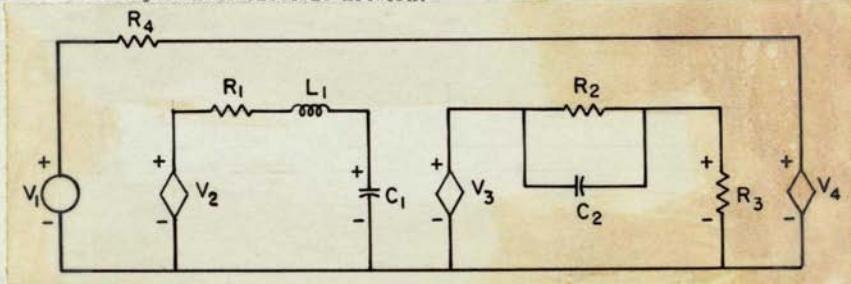


Fig. 3.14

where  $V_1$  is equivalent to  $r(t)$

$V_2$  is equivalent to  $e(t)$

$V_3$  is equivalent to  $c(t)$

$V_4$  is equivalent to  $b(t)$

## III D2 Control System Model and Its Step Response

Using a control system design problem adapted from D'Azzo and Houpis, pp. 408-411 [DA-1] we illustrate modeling of the system and use of NASAP to tabulate and plot the step response.

Consider the unity feedback control system with cascade lead compensation shown in Fig. 3.15. To determine the step response when  $K = 10$  we first obtain the NASAP

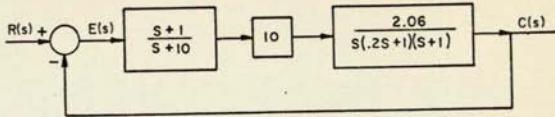


Fig. 3.15

circuit model shown in Fig. 3.16. In this model

V5 represents the input  $R(s)$

V4 represents the output  $C(s)$

V6 represents 2.06 times the error signal;  $2.06 E(s)$

Note that the system gain,  $K = 10$ , is included in the dependency relation of V1.

Also in Fig. 3.16 we have NASA output, namely the transfer function zeros and poles based on a simple flowgraph having only right first order loops. The desired step response is given in Fig. 3.17. Figure 3.17a is the response function and 3.17b gives 51 discrete time values from zero to six seconds in equal increments. Finally in Fig. 3.18 we have the plot of this time response.

These NASA outputs can be used to determine the key step response characteristics. Note in particular that the steady state value is unity since Fig. 3.17a we see that the residue of the pole at the origin is unity. Furthermore in Fig. 3.17b the peak overshoot occurs at  $t = 1.44$  seconds and is 9.15%. This response settles (for 2% tolerance) in 2.04 seconds.

NASAP MODEL FOR CONTROL SYSTEM

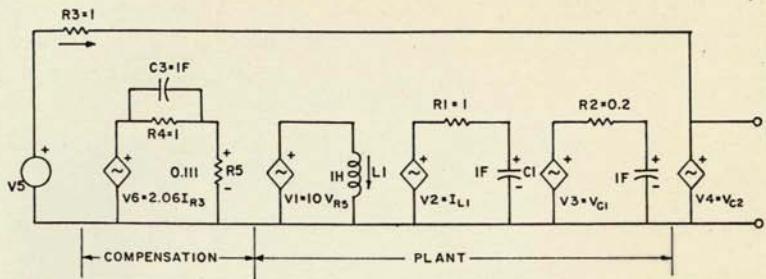


Fig. 3.16

Resistance Values in Ohms

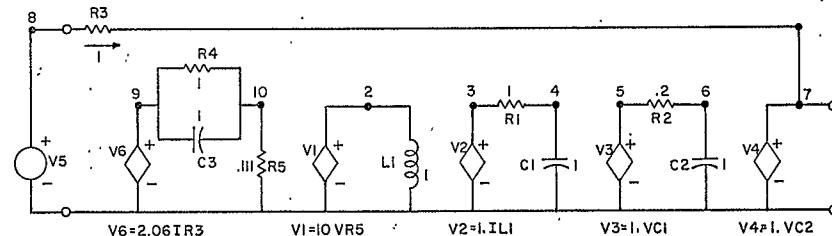
$V_5$  represents the input  $R(s)$

$V_4$  represents the output  $C(s)$

$V_6$  represents 2.06 times the error signal; 2.06  $E(s)$

NASAP PR, RLFM LEAD ~~COMPENSATION~~ COMPENSATION

V1 1 2 1.0 VR5  
 L1 2 1 1.0 H  
 V2 1 3 1.0 IL1  
 R1 1 1  
 C1 4 1 1  
 V3 5 1.0 VL1  
 R2 6 9 0.2  
 C2 6 1 LF  
 V4 7 8 1.0 VR2  
 R3 8 7 1.0  
 V5 9 8 1.0  
 V6 1 9 >0.06 .R3  
 C3 9 10 F  
 P4 9 10 1.000  
 P5 10 1 0.111  
 B01P01  
 VV4/VV5  
 FREQ 1.0 1.0 0.05  
 TIME 6.0  
 EXECUTE



NUMBER OF LC.JPS PER ORDER

1 = 8  
 2 = 5  
 3 = 2

~~83~~ TRANSFER FUNCTION VV4/VV5

$$(1.00E 00 +1.00E 00 S)$$

$$H(S) = 1.030E 02 *$$

$$(1.03E 02 +1.55E 02 S +6.51E 01 S^2 +1.60E 01 S^3 +1.00E 00 S^4)$$

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

ZERO OF TRANSFER FUNCTION

$$1 -0.17964E 01 0.24072E 01$$

ZERO REAL PART IMAG. PART

$$2 -0.17968E 01 -0.24072E 01$$

1 -0. 0000E 01 0.0000E 00

$$3 -0.1000E 01 -0.48388E -11$$

$$4 -0.1141E 02 0.80302E -09$$

STEP	RESPONSE FUNCTION	TIME	RESPONSE
F(T) =		VV4/VV5	
(-0.4541E 00 J 0.5566E 00 ) E	(-0.1797E 01 J 0.2407E 01 ) T	0.0000E 00	0.00000000E 00
		0.1200E 00	0.19343615E-01
		0.2400E 00	0.10478659E 00
		0.3600E 00	0.24626612E 00
		0.4800E 00	0.41497654E 00
		0.6000E 00	0.58518551E 00
		0.7200E 00	0.73095842E 00
		0.8400E 00	0.86009700E 00
		0.9600E 00	0.96466139E 00
		0.1080E 01	0.10794307E 01
		0.1200E 01	0.10695372E 01
		0.1320E 01	0.10684190E 01
		0.1440E 01	0.10919455E 01 peak ← 9%
		0.1560E 01	0.10848783E 01 overshoot
		0.1680E 01	0.10101990E 01
		0.1800E 01	0.10539808E 01
		0.1920E 01	0.10378954E 01 settling
		0.2040E 01	0.10233526E 01 time > 0.10233526E 01 ← 2% tolerance
		0.2160E 01	0.10112232E 01
		0.2280E 01	0.10026340E 01
		0.2400E 01	0.99558346E 00
		0.2520E 01	0.99500891E 00
		0.2640E 01	0.99141634E 00
		0.2760E 01	0.99127169E 00
		0.2880E 01	0.99309821E 00
		0.3000E 01	0.99345618E 00
		0.3120E 01	0.99502158E 00
		0.3240E 01	0.99655831E 00
		0.3360E 01	0.99791598E 00
		0.3480E 01	0.99901557E 00
		0.3600E 01	0.99983203E 00
		0.3720E 01	0.100004805E 01
		0.3840E 01	0.100006971E 01
		0.3960E 01	0.100008297E 01
		0.4080E 01	0.100009280E 01
		0.4200E 01	0.106004200E 01
		0.4320E 01	0.100006075E 01
		0.4440E 01	0.100004568E 01
		0.4560E 01	0.100004109E 01
		0.4680E 01	0.100001831E 01
		0.4800E 01	0.100000811E 01
		0.4920E 01	0.100000067E 01
		0.5040E 01	0.39995738E 00
		0.5160E 01	0.39992949E 00
		0.5280E 01	0.39991864E 00
		0.5400E 01	0.39991995E 00
		0.5520E 01	0.39991929E 00
		0.5640E 01	0.39999422E 00
		0.5760E 01	0.399995673E 00
		0.5880E 01	0.99997056E 00
		0.6000E 01	0.99998248E 00

	STEP	RESPONSE
	-9.00E-01	-7.00E-01 -5.00E-01 -3.00E-01 -1.00E-01 1.00E-01 3.00E-01 5.00E-01 7.00E-01 9.00E-01 1.10E 00
	0.0000E 00	* * * * *
		*
		*
		*
	6.0000E-01	* * * *
		*
		*
		*
	1.2000E 00	* * * *
		*
		*
		*
	1.8000E 00	* * * *
		*
		*
		*
	2.4000E 00	* * * *
		*
		*
		*
	3.0000E 00	* * * *
		*
		*
		*
18	3.6000E 00	* * * *
		*
		*
		*
	4.2000E 00	* * * *
		*
		*
		*
	4.8000E 00	* * * *
		*
		*
		*
	5.4000E 00	* * * *
		*
		*
		*
	6.0000E 00	* * * *
		*
		*
		*

CHAPTER IV  
CONTROL SYSTEM ANALYSIS IN THE FREQUENCY DOMAIN

IVA ANALYSIS OBJECTIVES

As mentioned in Chapter III control system design is usually aimed at a set of specifications. Therefore we start with a summary of typical performance specifications as general background. For convenience we adapt the tabulations given by Grabbe, Ramo, and Woodruff [GA 2] and list some of them for transient response in Table 4.1a and for frequency response in Table 4.1b. These are sufficient to indicate the sort of specifications that can be expected for the class of linear time-invariant single input single output control systems being considered in this manual.

There are several approximations or rules of thumb which were developed by control engineers for use when time or facilities are not available for a more exact analysis of this class of systems. They are also useful as rough checks on the results of a computer analysis. The more common of these rules of thumb are presented, virtually unchanged from [GA 2] in Table 4.2. They must, however, be used with caution since being approximations, they do not apply with equal validity to all control systems. Note that the approximations for transient response are applicable only for step inputs.

Following Table 4.2 we have Fig. 4.1 which shows typical step and frequency response curves to help pin down the definition of some of the terms used in these tables.

Table 4.1a

COMMON PERFORMANCE SPECIFICATIONS		Transient response
Type	Definition	General Remarks
1. Transient overshoot	Usually taken as ratio of peak of transient to final value for a step <i>input</i> .	Convenient when transient solution is available. Can be estimated from root locus or frequency response. Useful for nonlinear systems. System must be excited by step inputs and be underdamped.
2. Settling time	Defined as time to reach and remain within a specified percentage of final value (often as 5% or 2%) after a step <i>input</i> .	See 1 above. Used for systems which require rapid synchronization, e.g., fire control system.
3. Steady-state error	Final error existing between desired and actual output.	See 1 above. Easily calculated from static characteristics or final value theorem. Useful when input is simple aperiodic function. Can include frequency components which arise in nonlinear systems.
4. Rise time	Defined as (a) time to $\frac{1}{2}$ the final value, or (b) slope at $\frac{1}{2}$ the final value, or (c) time between 10% and 90% of final value after a step <i>input</i> .	Easily estimated from frequency response or root locus and is indicative of band pass of system. Used for overdamped systems. Has found application in process controls where characteristics (1) or (2) may not be easily recognized.
5. Dead time	Defined as (a) time for output action to be initiated, or (b) for output to reach a given level (10% or 50%), or (c) time to the intersection of the slope of the transient at $\frac{1}{2}$ the final value and the initial value after a step input.	See 1 above. Easily estimated from frequency response and is indicative of phase shift near gain crossover in systems. Useful when delay times exist in system. Used for overdamped systems. Both rise and delay time derive from filter theory.
6. Absolute damping decrement factor	Defined as the real part of the roots of a quadratic system and as such determines the rate of decay of transient.	Convenient method of interpreting more complex systems in terms of quadratic systems. Valuable in combination with relative damping in work with root locus analysis. Has had extensive use in systems demanding prescribed transient performance, particularly when the time decay is important, e.g., in autopilots.
7. Damping ratio	Damping ratio is defined as $\zeta$ in the quadratic $s^2 + 2\omega_n s + \omega_n^2$ and indicates the decay per cycle of the natural frequency.	Useful because it is a parameter in nondimensional plot of quadratic response. Used in combination with 6 above in root locus analysis. Used when number and size of overshoot are important. In combination with 6 above defines decay of oscillatory component of transient.

Table 4.1b

COMMON PERFORMANCE SPECIFICATIONS		Frequency response.
8. Phase margin	Defined as $180^\circ + \text{phase shift}$ at unity gain of the open loop frequency response.	Used as a rule of thumb in frequency response analysis to indicate stability and performance. Easy to use and to obtain directly from frequency response diagram.
9. Gain margin	Gain margin is ratio of maximum stable gain to actual gain, i.e., gain at phase crossover.	Same as 8. Indicates relative sensitivity of system to gain variations. Can be calculated by Routh's criterion. Not as good a criterion for performance as 3. Little used.
10. $M_m$ peak	Ratio of maximum of closed loop frequency response to a low frequency value.	Used with Nyquist and frequency response analysis. Rules of thumb relate $M_m$ and transient overshoot. Easy to calculate from frequency response diagram.
11. Band width	Defined variously (a) usually as frequency where closed loop response falls to $\sqrt{\frac{1}{2}}$ or 3 db of its low frequency value, or (b) sometimes as the frequency at the significant peak $M_m$ , or (c) the crossover of the open loop response.	Used with frequency response analysis and is related to speed of response of system. Used also when definite frequency bandpass is needed for fidelity. $M_m$ , bandpass, and the phase shift at these values give a good indication of the closed loop response and are often used when a number of closed loops are operated in tandem as system.
12. Static error coefficient	Defined as the final error resulting from a continuous input of position, or velocity, or acceleration, etc. The magnitude of the input and the maximum tolerable error must be specified.	Used to set low-frequency gain of open loop frequency response. Useful where steady inputs are encountered.
13. Dynamic error coefficients (or steady-state error coefficients)	Defined as the steady-state error resulting from the derivatives of the input function. The time function and/or its derivatives must be specified as well as the maximum tolerable error.	Relates system gain and time constants to errors arising from higher derivatives of input. Used to estimate error resulting from varying input to given system and conversely to determine closed loops pole-zero location to give desired error. Accurate where input varies at slow rate compared to bandpass. Becomes poorer as input varies more rapidly because of transient effects.

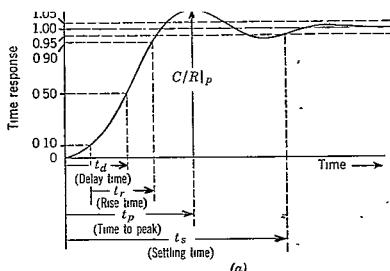
Table 4.2

Parameter	Approximation	Remarks
Time to peak	$t_p \approx \pi/\omega_c$ where $t_p$ = time from step input to peak value of response transient, seconds $\omega_c$ = open loop crossover frequency, radians/second	Increases with a dominant complex pair of closed loop poles, the open loop crossover frequency, $\omega_c$ , times the time to peak, $t_p$ , is about 3 or $\pi$ . In other words, the time to peak is about half the period corresponding to the open loop crossover frequency.
Peak overshoot	$C/R _p \approx 0.85M_m$ where $C/R _p$ = peak value of transient response to a step input $M_m$ = maximum value of closed loop frequency response	The peak value of the transient response, $C/R _p$ , to a unit step input is generally less than the maximum steady-state value, $M_m$ , of the closed loop frequency response. The maximum value of $C/R _p$ generally approaches 2.0 while the maximum value of $M_m$ approaches infinity. For many applications "good" servos are those with the values of $M_m$ between 1.3 and 1.5.
Damping ratio	$\xi = 1/(2M_s)$ where $\xi$ = damping ratio $M_s$ = value of closed loop frequency response at the corner frequency	The damping ratio may be approximated from the value of the closed loop frequency response of the system at the corner frequency, $\omega_c$ (the frequency at which the lines asymptotic to the log magnitude curve intersect). This is exact for a second order system.
Settling time	$t_s(5\%) \approx 3\sqrt{1 - \xi^2}/\zeta\omega_d$ $t_s(2\%) \approx 5\sqrt{1 - \xi^2}/\zeta\omega_d$ $t_s(5\%) \approx 3T_{eq}$ where $t_s$ = time for response to step input to settle to within some per cent of final value, seconds $T_{eq}$ = time for response to reach 63% of final value $\omega_d$ = damped natural frequency, radians/second $\zeta$ = damping ratio	The settling time, $t_s$ , is generally defined as the time for the system to settle to within 5 or sometimes 2% of the final value. In either case it is quite difficult to predict $t_s$ for an underdamped system because it is subject to fluctuations of about one-half the period of oscillation for only small changes in system parameters.

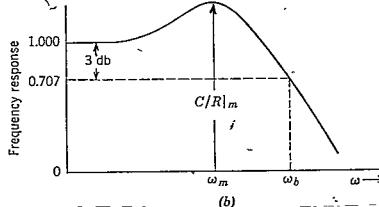
Table 4.2

## RULE-OF-THUMB APPROXIMATIONS (Continued)

Parameter	Approximation	Remarks
Rise time	$t_r \omega_0 \approx t_r \omega_m \approx 1.3$ where $t_r$ = rise time (10 to 90%) $\omega_t$ = (defined above) $\omega_m$ = (defined above)	The system's rise time, $t_r$ , which is here considered to be the time for the response to a step input to go from 10 to 90% of its final value may be approximated as indicated for systems with a $M_m$ value of about 1.3 to 1.5.
Phase margin at crossover frequency	$\gamma_c \geq 40^\circ$ where $\gamma_c$ = open loop phase margin at the crossover frequency	A phase margin of $40^\circ$ at the unity gain (crossover) frequency generally corresponds to a $M_m$ ratio of approximately 1.5. Since this value of $M_m$ is the maximum ordinarily considered feasible, the phase margin should be $40^\circ$ or greater.
Oscillation frequency	$\omega_t \approx \omega_m \approx 0.75\omega_c$ where $\omega_t$ = oscillation frequency of transient response, radians/second $\omega_m$ = frequency at which $M_m$ occurs, radians/second $\omega_c$ = open loop gain crossover frequency, radians/second	The frequency of oscillation of the transient response, $\omega_t$ , is generally about equal to the frequency, $\omega_m$ , at which the frequency response peak, $M_m$ , occurs. Both $\omega_m$ and $\omega_t$ are usually less than $\omega_c$ , the open loop crossover frequency. For the "good" servos with $M_m = 1.3$ to 1.5 an approximate relationship is as indicated. In this approximation $\omega_t$ is used to mean essentially the same thing as $\omega_d$ , the damped natural frequency, previously defined for a system with a dominant complex pair of poles.



(a)



(b)

Figure 4.1. (a) Typical servo system response to unit step input.  
 (b) Typical system frequency response (closed loop).

#### IVB BODE AND ROOT LOCUS PLOTS

Bode showed that the phase angle of a (minimum phase) transfer function  $G$  could be related to the rate at which the magnitude of  $G$  decreases with increasing frequency. This is the basis for the "frequency response" method of analysis. In this method, the magnitude of  $G$  in decibels and the phase angle, are plotted on semilog paper as functions of the frequency  $\omega$  (plotted on the log scale) with  $j\omega$  substituted for  $s$  in  $G$ . The value of  $|G|$  in decibels (dB) can be found from the value of  $|G|$  by the following equation.

$$|G|_{\text{dB}} = 20 \log_{10} |G| \quad (4.1)$$

For preliminary estimation of compensating networks to insure stability or improve performance, it is often possible to omit the phase angle plot and to make use of an approximate attenuation plot (or plot of  $|G|$ ). Exact plots are needed for a final check after selection of the proposed stabilizing transfer functions, however. The method of drawing approximate attenuation plots is described in most textbooks on control theory. Before we discuss and illustrate how NASAP obtains exact plots of the magnitude and phase of  $G$ , we provide a collection of typical transfer functions with their corresponding Bode plots in Table 4.3. These plots indicate the gain and phase margins.

Another important tool for analysis and synthesis of linear control systems, usually attributed to Evans, is known as the "Root Locus" method. As with the frequency response methods, its importance derives from the fact that it helps provide insight into the significant aspects of any particular system. It is not restricted to direct feedback systems nor to systems with open loop poles and zeros in the left half plane.

It is recalled that for any closed-loop system with "degenerative feedback" see Fig. 4.2, the closed-loop transfer function is

$$\frac{C}{R} = \frac{G}{1 + GH} \quad (4.2)$$

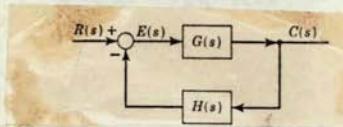


Fig. 4.2: Block diagram of a basic feedback control system.

The closed-loop response is determined by the roots of the denominator, i.e., the characteristic equation. The roots must all be in the left half plane in order that the system be stable. Furthermore the time-domain design parameters, such as peak time maximum overshoot, damping factor, and settling time, are initially related to the s-plane location of the roots of the characteristic equation of the control system, which are the poles of the closed-loop system function. Accordingly, a knowledge of how the roots of  $1 + GH$  vary when the gain constant of  $GH$  varies should be of considerable assistance in understanding the system. A plot of the locus of the characteristic roots of the control system with the system loop gain as a parameter is commonly known as the root locus.

The design of feedback control systems by use of the root-locus method involves the reshaping of the root-locus plots by shifting or introducing open-loop poles and zeros. As a preliminary to the discussion of the design aspects, the effects of shifting open-loop poles and zeros are first indicated with the aid of examples in Table 4.3.

Gain Adjustment. When the preliminary analysis of a control system indicates that the system is unstable or that the over-all performance is inadequate, steps must be taken to improve the system performance. The most direct and simplest way of changing the performance is by the adjustment of the system gain.

Table 4.3

## PLOTS FOR TYPICAL TRANSFER FUNCTIONS

$G(s)$	Bode diagram	Root locus	Comments
$\frac{K}{s\tau_1 + 1}$	<p>Phase margin: <math>\frac{1}{\tau_1} \text{ db/oct}</math></p>	<p>Root locus: <math>\frac{1}{\tau_1}</math></p>	Stable, gain margin = $\infty$
$\frac{K}{(s\tau_1 + 1)(s\tau_2 + 1)}$	<p>Phase margin: <math>-12 \text{ db/oct}</math></p>	<p>Poles: <math>r_1, r_2</math></p>	Stable, gain margin = $\infty$
$\frac{K}{(s\tau_1 + 1)(s\tau_2 + 1)(s\tau_3 + 1)}$	<p>Gain margin: <math>-12 \text{ db/oct}</math></p> <p>Phase margin: <math>-18 \text{ db/oct}</math></p>	<p>Poles: <math>r_1, r_2, r_3</math></p>	Unstable, shown; can be made stable by reducing gain
$\frac{K}{s}$	<p>Phase margin: <math>-90^\circ</math></p>		Ideal integrator; stable

(Continued)

Table 4.3 (con't.)

$G(s)$	Bode diagram	Root locus	Comments
$\frac{K}{s(sr_1 + 1)}$	<p>Phase margin: <math>-180^\circ</math></p> <p>Magnitude: <math>0 \text{ db}</math>, <math>-\frac{1}{r_1} \text{ rad/s}</math>, <math>-12 \text{ db/oct}</math></p>	<p>Poles: <math>r_1</math></p> <p>Zeros: <math>-\frac{1}{r_1}</math></p>	Stable, gain margin = $\infty$
$\frac{K}{s(sr_1 + 1)(sr_2 + 1)}$	<p>Phase margin: <math>-180^\circ</math></p> <p>Magnitude: <math>0 \text{ db}</math>, <math>\frac{1}{r_1}, \frac{1}{r_2} \text{ rad/s}</math>, <math>-12 \text{ db/oct}</math></p> <p>Gain margin: <math>0</math></p>	<p>Poles: <math>r_1, r_2</math></p> <p>Zeros: <math>-\frac{1}{r_1}, -\frac{1}{r_2}</math></p>	System is stable; becomes unstable with increased gain
$\frac{K(sr_a + 1)}{s(sr_1 + 1)(sr_2 + 1)}$	<p>Phase margin: <math>-180^\circ</math></p> <p>Magnitude: <math>0 \text{ db}</math>, <math>\frac{1}{r_1}, \frac{1}{r_2}, -6 \text{ rad/s}</math>, <math>-12 \text{ db/oct}</math></p>	<p>Poles: <math>r_1, r_2</math></p> <p>Zeros: <math>-\frac{1}{r_1}, -\frac{1}{r_2}, -\frac{1}{r_a}</math></p>	Above control system with phase-lead (derivative) compensator; stable
$\frac{K}{s^2}$	<p>Gain margin = 0</p> <p>Phase margin = 0</p> <p>Magnitude: <math>0 \text{ db}</math>, <math>\omega = 1/r_1 \text{ rad/s}</math>, <math>-12 \text{ db/oct}</math></p>	<p>Double pole</p> <p>Poles: <math>r_1, r_2</math></p>	Inherently unstable; must be compensated

(Continued)

Table 4.3 (con't.)

$G(s)$	Bode diagram	Root locus	Comments
$\frac{K}{\sqrt{(sr_1 + 1)}}$			Inherently unstable must be compensated
$\frac{K(sr_a + 1)}{s^2(sr_1 + 1)}$			Stable for all gains
$\frac{K(sr_a + 1)(sr_b + 1)}{(sr_1 + 1)(sr_2 + 1)(sr_3 + 1)(sr_4 + 1)}$			Conditionally stable; stable at low gain, becomes unstable as gain is raised, again becomes stable as gain is further increased, and becomes unstable for very high gains
$\frac{K(sr_a + 1)}{s^2(sr_1 + 1)(sr_2 + 1)}$			Conditionally stable; becomes unstable at high gain

Adapted from George J. Thaler and Robert G. Brown, *Analysis and Design of Feedback Control Systems*, 2nd Ed., McGraw-Hill, New York, 1960.

However, for most control systems the design specifications cannot be met by gain adjustment alone. The usual alternative is the introduction of compensating devices into the control system.

The adequate gain setting for a control system can be determined from the gain-phase plot of the system or from the Bode diagram of the system.

A change in system gain usually affects practically all of the system design parameters. For instance, an increase in system gain may cause a reduction of the system error, may increase the speed of response of the control system, and may make the system more oscillatory. The effects of gain variations upon the behavior of a control system are conveniently observed on the root locus plot of the control system also.

#### NASAP Output

For a given transfer function  $G$  with all the coefficients known, the frequency response is calculated by setting  $s = j\omega$  and simplifying the expression to a linear combination of real and imaginary terms:

$$G = A(\omega) + jB(\omega) . \quad (4.3)$$

The magnitude of  $G$  and the angle  $\theta(\omega)$  are computed according to the equations,

$$|G(j\omega)| = \sqrt{A^2(\omega) + B^2(\omega)} , \quad (4.4)$$

and

$$\theta(\omega) = \tan^{-1} \frac{B(\omega)}{A(\omega)} . \quad (4.5)$$

Now if  $\omega$  is made to vary, then for each value of  $\omega$  the  $|G(j\omega)|$  and  $\theta(\omega)$  can be obtained over the frequency range of interest and thus can be made available for plotting. With the complex arithmetic capability of FORTRAN IV, these computations are easily done in NASAP. The Bode plot consists of the  $|G(j\omega)|$  in decibel units and  $\theta(\omega)$  in degrees versus the  $\log_{10}\omega$ , taken over the frequency range specified

by the user.

To obtain a root locus plot for a control system, Fig. 4.2, it is required to find the values of  $s$  for which  $GH = -1$  (or  $1 + GH = 0$ ). For this it is necessary that the angle of the complex number,  $GH$ , be 180 degrees and the magnitude of  $GH$  be unity. Thus, the complex number,  $s$ , must be selected so that the angle of the complex number,  $GH$ , is 180 degrees. When such a complex number for  $s$  is determined, a value of gain  $K$  can then be found which will make the magnitude of  $GH$  unity, although this value of  $K$  might not necessarily be the same as the value specified in the transfer function. However, after a locus of values of  $s$  for which  $GH = 180$  degrees has been found, somewhere along this locus one can find a number that yields  $|GH| = 1$  for the specified value of  $K$ . NASA furnishes the necessary data in tabulated form to obtain the locus of points for which  $GH = 180$  degrees. To obtain the root locus plot directly by the computer requires an extra program that will not be described here. Such programs are available in the literature as illustrated by Program D91RTL by Vernon [Appendix I in VE 1] and another by Krall and Fornaro [see KR1 or KR2].

An alternative approach is to use the root sensitivity data available from NASA to approximate the root locus plot. Such sensitivity data for an aerospace control problem is given in Chapter VI.

As an illustration of the open loop Bode plot output of NASA we use a control system plant whose transfer function has a seventh degree polynomial denominator. The NASA print out is shown in Fig. 4.3. This Eisenberg control problem [EI 1] is discussed further in Chapters V and VII.

## NASAP PROBLEM EISENBERG CONTROL SYSTEM

RADIAN  
NONE

V1 1 2 1.

R1 2 3 6.

C1 3 1 1F

V2 1 4 1. VC1

R2 4 5 2.

C2 5 1 1F

V3 1 6 1. VC2

R3 6 7 2.

C3 7 1 1F

V4 1 8 1. VC3

R4 8 9 2.

C4 9 1 1F

V5 1 10 1. VC4

R5 10 11 1.

C5 11 1 1F

V6 1 12 1. VC5

R6 12 13 1.

C6 13 1 1F

V7 1 14 1. VC6

R7 14 15 1.

C7 15 1 1F

V8 1 16 1. VC7

OUTPUT

VV8/VV1/V6

FREQ -2.0 1.0 0.05

EXECUTE

NUMBER OF LOOPS PER ORDER

1= 8

2= 21

3= 35

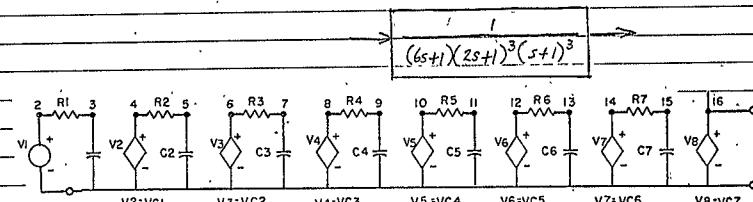
4= 35

5= 21

6= 7

7= 1

72.8



TRANSFER FUNCTION VV8/VV1/V6

( 1.00E 00 )

H(S)= 2.083E-02\*

( 2.00E-02 +3.125E-01 S +1.61E 00 S 2 +5.44E 00 S 3 +9.25E 00 S 4 +9.00E 00 S 5 +4.67E 00 S 6 +1.00E 00 S 7 )

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

ZERO OF TRANSFER FUNCTION

NONE

1	-0.97827E 00	-0.17632E-02
2	-0.97827E 00	0.17632E-02
3	-0.51017E 00	-0.93019E-02
4	-0.51017E 00	0.93019E-02
5	-0.48950E 00	-0.11737E-02
6	-0.48950E 00	0.11737E-02
7	-0.16667E 00	0.21177E-12

	LGD(FREQ)	FREQ	20.*LGND(ABS(H))	PHI(H)	ABS(H)	LGD(ABS(H))	Phase factor ≈ (15° - 6.5°) ..	
1	-0.1999599E-01	0.1000002E-01	-0.2118707E-01	-0.8589749E-01	0.9974567E-00	-0.115935E-02	Phase factor ≈ (15° - 6.5°) ..	
2	-0.1949799E-01	0.1122021E-01	-0.2783848E-01	-0.9636519E-01	0.9968001E-00	-0.1391942E-02	Phase factor ≈ (15° - 6.5°) ..	
3	-0.1899599E-01	0.1250929E-01	-0.3502743E-01	-0.1081046E-02	0.9959754E-00	-0.1751372E-02	Phase factor ≈ (15° - 6.5°) ..	
4	-0.1849799E-01	0.1412543E-01	-0.4476763E-01	-0.1212687E-02	0.9949384E-00	-0.223818E-02	Phase factor ≈ (15° - 6.5°) ..	
5	-0.1799599E-01	0.1584996E-01	-0.6584931E-01	-0.1360279E-02	0.9936363E-00	-0.2775545E-02	Phase factor ≈ (15° - 6.5°) ..	
6	-0.1749599E-01	0.1776238E-01	-0.6975049E-01	-0.1525727E-02	0.9920018E-00	-0.3447526E-02	Phase factor ≈ (15° - 6.5°) ..	
7	-0.1699599E-01	0.1995267E-01	-0.8771539E-01	-0.1711147E-02	0.9909522E-00	-0.4381679E-02	Phase factor ≈ (15° - 6.5°) ..	
8	-0.1649999E-01	0.2236720E-01	-0.11120279E-00	-0.1919886E-02	0.9873838E-00	-0.5513981E-02	Phase factor ≈ (15° - 6.5°) ..	
9	-0.1599599E-01	0.2511059E-01	-0.1386055E-00	-0.2151541E-02	0.9841697E-00	-0.6930023E-02	Phase factor ≈ (15° - 6.5°) ..	
10	-0.1549799E-01	0.2814388E-01	-0.17142154E-00	-0.2411990E-02	0.9801527E-00	-0.8766272E-02	Phase factor ≈ (15° - 6.5°) ..	
11	-0.1499599E-01	0.3162285E-01	-0.21836347E-00	-0.2703303E-02	0.9751429E-00	-0.1093174E-01	Phase factor ≈ (15° - 6.5°) ..	
12	-0.1449599E-01	0.3545143E-01	-0.2743355E-00	-0.3042916E-02	0.9689098E-00	-0.1371675E-01	Phase factor ≈ (15° - 6.5°) ..	
13	-0.1400000E-01	0.3981075E-01	-0.3439497E-00	-0.3393636E-02	0.9611753E-00	-0.1719739E-01	Phase factor ≈ (15° - 6.5°) ..	
14	-0.1349599E-01	0.4466484E-01	-0.4379479E-00	-0.3799130E-02	0.9516124E-00	-0.2133993E-01	Phase factor ≈ (15° - 6.5°) ..	
15	-0.1299599E-01	0.5011881E-01	-0.5389174E-00	-0.4251649E-02	0.9398410E-00	-0.2654562E-01	Phase factor ≈ (15° - 6.5°) ..	
16	-0.1245449E-01	0.5623242E-01	-0.5734144E-00	-0.4729144E-02	0.9254282E-00	-0.3365724E-01	Phase factor ≈ (15° - 6.5°) ..	
17	-0.1200000E-01	0.6309575E-01	-0.63722759E-00	-0.5311436E-02	0.9078959E-00	-0.4196395E-01	Phase factor ≈ (15° - 6.5°) ..	
18	-0.1150000E-01	0.7079464E-01	-0.6441428E-01	-0.5939394E-02	0.8867330E-00	-0.5220710E-01	Phase factor ≈ (15° - 6.5°) ..	
19	-0.1090000E-01	0.7942930E-01	-0.6936552E-01	-0.6618787E-02	0.8614247E-00	-0.6470262E-01	Phase factor ≈ (15° - 6.5°) ..	
20	-0.1040000E-01	0.8912522E-01	-0.7628631E-01	-0.7315389E-02	0.8314877E-00	-0.8014417E-01	Phase factor ≈ (15° - 6.5°) ..	
21	-0.1000000E-01	0.1000000E-01	-0.9760276E-01	-0.8022527E-02	0.7965235E-00	-0.98944117E-01	Phase factor ≈ (15° - 6.5°) ..	
22	-0.9500000E-01	0.1122016E-00	-0.9742339E-00	-0.9109307E-02	0.75662759E-00	-0.1213197E-00	Phase factor ≈ (15° - 6.5°) ..	
23	-0.9000000E-01	0.1259232E-00	-0.9852665E-01	-0.1009092E-02	0.7107000E-00	-0.1453133E-00	Phase factor ≈ (15° - 6.5°) ..	
24	-0.8500000E-01	0.1415237E-00	-0.9604797E-01	-0.1117293E-02	0.6600246E-00	-0.1843938E-00	Phase factor ≈ (15° - 6.5°) ..	
25	-0.8000000E-01	0.1504482E-00	-0.9367194E-01	-0.1233394E-02	0.6047881E-00	-0.2133937E-00	Phase factor ≈ (15° - 6.5°) ..	
26	-0.7500000E-01	0.1778245E-00	-0.1253347E-01	-0.1320402E-02	0.5548602E-01	-0.2629185E-00	Phase factor ≈ (15° - 6.5°) ..	
27	-0.7000000E-01	0.2175952E-00	-0.1492426E-01	-0.1494418E-02	0.4044180E-01	-0.3147779E-00	Phase factor ≈ (15° - 6.5°) ..	
28	-0.6500000E-01	0.2239718E-00	-0.1749592E-01	-0.1635350E-02	0.4218943E-01	-0.3747793E-00	Phase factor ≈ (15° - 6.5°) ..	
29	-0.6000000E-01	0.2511834E-00	-0.1878163E-01	-0.1787576E-02	0.3599887E-02	-0.4464033E-00	Phase factor ≈ (15° - 6.5°) ..	
30	-0.5500000E-01	0.2814375E-00	-0.1945691E-02	-0.1651522E-02	0.3000539E-00	-0.5522804E-00	Phase factor ≈ (15° - 6.5°) ..	
31	-0.5000000E-01	0.3162271E-00	-0.225335E-02	-0.1482119E-02	0.2439737E-00	-0.6126574E-00	Phase factor ≈ (15° - 6.5°) ..	
32	-0.4500000E-01	0.3549412E-00	-0.1492787E-02	-0.1304739E-02	0.1930300E-00	-0.7134738E-00	Phase factor ≈ (15° - 6.5°) ..	
33	-0.4000000E-01	0.3918006E-00	-0.1657784E-02	-0.111V113E-02	0.1482802E-00	-0.8219167E-00	Phase factor ≈ (15° - 6.5°) ..	
34	-0.3500000E-01	0.4466857E-00	-0.1912375E-02	-0.9292380E-02	0.1103600E-00	-0.9571081E-00	Phase factor ≈ (15° - 6.5°) ..	
35	-0.3000000E-01	0.5011859E-00	-0.2119399E-02	-0.7333747E-02	0.7942329E-01	-0.1089998E-01	Phase factor ≈ (15° - 6.5°) ..	
36	-0.2500000E-01	0.5623240E-00	-0.2510066E-02	-0.5343427E-02	0.5520728E-01	-0.1250036E-01	Phase factor ≈ (15° - 6.5°) ..	
37	-0.2100000E-01	0.6250000E-00	-0.2933232E-02	-0.3323232E-02	0.3761113E-01	-0.1431664E-01	Phase factor ≈ (15° - 6.5°) ..	
38	-0.2000000E-01	0.6309502E-00	-0.3263536E-02	-0.3323232E-02	0.3239208E-01	-0.1621222E-01	Phase factor ≈ (15° - 6.5°) ..	
39	-0.1900000E-01	0.3076433E-00	-0.3242444E-02	-0.1305595E-02	0.2392086E-01	-0.1826686E-01	Phase factor ≈ (15° - 6.5°) ..	
40	-0.1800000E-01	0.3945224E-00	-0.3465337E-02	-0.1696626E-02	0.1490436E-01	-0.2047808E-01	Phase factor ≈ (15° - 6.5°) ..	
41	-0.1700000E-01	0.4891248E-00	-0.3904619E-02	-0.2656567E-02	0.8987598E-02	-0.2047808E-01	Phase factor ≈ (15° - 6.5°) ..	
42	-0.1600000E-01	0.5999979E-00	-0.4999761E-02	-0.4584215E-02	0.5198841E-02	-0.2284095E-01	Phase factor ≈ (15° - 6.5°) ..	
43	-0.1500000E-01	0.6999974E-00	-0.5961868E-02	-0.5458421E-02	0.3779908E-02	-0.2799085E-01	Phase factor ≈ (15° - 6.5°) ..	
44	-0.1499242E-01	0.7112201E-00	-0.6059620E-02	-0.6209105E-02	0.1588223E-02	-0.3075787E-01	Phase factor ≈ (15° - 6.5°) ..	
45	-0.1499347E-01	0.7999984E-00	-0.6151573E-02	-0.6989037E-02	0.8389707E-03	-0.3037578E-01	Phase factor ≈ (15° - 6.5°) ..	
46	-0.1499787E-00	0.11412533E-00	-0.6147180E-02	-0.74217705E-03	0.44217705E-03	-0.3363741E-01	Phase factor ≈ (15° - 6.5°) ..	
47	-0.1499999E-00	0.1584809E-00	-0.6172748E-02	-0.1147180E-02	0.2179228E-03	-0.3661691E-01	Phase factor ≈ (15° - 6.5°) ..	
48	-0.1499999E-00	0.1778272E-00	-0.6283331E-02	-0.1294729E-02	0.2179228E-03	-0.3661691E-01	Phase factor ≈ (15° - 6.5°) ..	
49	-0.1499999E-00	0.1995255E-00	-0.6396767E-02	-0.1441615E-02	0.1075415E-03	-0.3960339E-01	Phase factor ≈ (15° - 6.5°) ..	
50	-0.1499999E-00	0.3439987E-00	-0.6565311E-02	-0.1587636E-02	0.5216013E-04	-0.4232656E-01	Phase factor ≈ (15° - 6.5°) ..	
51	-0.1499999E-00	0.3999987E-00	-0.6826164E-02	-0.1613636E-02	0.2492653E-04	-0.4673337E-01	Phase factor ≈ (15° - 6.5°) ..	
52	-0.1499999E-00	0.42819369E-00	-0.6858633E-02	-0.1782925E-02	0.11764476E-04	-0.4929417E-01	Phase factor ≈ (15° - 6.5°) ..	
53	-0.1499999E-00	0.4499979E-00	-0.6985401E-02	-0.1726171E-02	0.54953358E-05	-0.5260005E-01	Phase factor ≈ (15° - 6.5°) ..	
54	-0.1499999E-00	0.3162264E-00	-0.7010520E-02	-0.1649730E-02	0.2465011E-05	-0.55943105E-01	Phase factor ≈ (15° - 6.5°) ..	
55	-0.1499999E-00	0.3248120E-00	-0.7118686E-02	-0.1649730E-02	0.11750401E-05	-0.5931664E-01	Phase factor ≈ (15° - 6.5°) ..	
56	-0.1499999E-00	0.3981058E-00	-0.7118633E-02	-0.1591741E-02	0.11750401E-05	-0.5931664E-01	Phase factor ≈ (15° - 6.5°) ..	
57	-0.1499999E-00	0.64966812E-00	-0.7254299E-02	-0.1491543E-02	0.9381847E-06	-0.6271493E-01	Phase factor ≈ (15° - 6.5°) ..	
58	-0.1499999E-00	0.5011844E-00	-0.7326270E-02	-0.14248479E-02	0.2435550E-06	-0.6613348E-01	Phase factor ≈ (15° - 6.5°) ..	
59	-0.1499999E-00	0.5622388E-00	-0.7391366E-02	-0.1371911E-02	0.11046513E-06	-0.6966822E-01	Phase factor ≈ (15° - 6.5°) ..	
60	-0.1499999E-00	0.6309549E-00	-0.14663232E-02	-0.13212326E-02	0.4993214E-07	-0.7316158E-01	Phase factor ≈ (15° - 6.5°) ..	
61	-0.1499999E-00	0.7079417E-00	-0.1529490E-02	-0.1294729E-02	0.2251877E-07	-0.7647449E-01	Phase factor ≈ (15° - 6.5°) ..	
62	-0.1499999E-00	0.8999977E-00	-0.1794320E-02	-0.1598827E-02	0.12393137E-03	-0.1013584E-07	-0.7984136E-01	Phase factor ≈ (15° - 6.5°) ..
63	-0.1499999E-00	0.8912466E-01	-0.1668299E-02	-0.1199103E-02	0.49359130E-08	-0.9341494E-01	Phase factor ≈ (15° - 6.5°) ..	
64	-0.1499999E-00	0.99999956E-01	-0.1737879E-02	-0.1166740E-02	0.2044577E-08	-0.0689394E-01	Phase factor ≈ (15° - 6.5°) ..	

LOG(FREQ) 20.\*LOG(ABS(H))  
 -2.00E\_02 -1.80E\_02 -1.60E\_02 -1.40E\_02 -1.20E\_02 -1.00E\_02 -8.00E\_01 -6.00E\_01 -4.00E\_01 -2.00E\_01 0.00E\_00  
 -2.0000E\_00 .....\*.....  
 . . . . .  
 . . . . .  
 -1.7500E\_00 .....\*.....  
 . . . . .  
 . . . . .  
 -1.5000E\_00 .....\*.....  
 . . . . .  
 . . . . .  
 -1.2500E\_00 .....\*.....  
 . . . . .  
 . . . . .  
 -1.0000E\_00 .....\*.....  
 . . . . .  
 . . . . .  
 -7.5000E\_01 .....\*.....  
 . . . . .  
 . . . . .  
 -5.0000E\_01 .....\*.....  
 . . . . .  
 . . . . .  
 -2.5000E\_01 .....\*.....  
 . . . . .  
 . . . . .  
 -9.5367E\_07 .....\*.....  
 . . . . .  
 . . . . .  
 2.5000E\_01 .....\*.....  
 . . . . .  
 . . . . .  
 5.0000E\_01 .....\*.....  
 . . . . .  
 . . . . .  
 7.5000E\_01 .....\*.....  
 . . . . .  
 . . . . .  
 1.0000E\_00 .....\*.....

LOG(FREQ) -2.00E 02 -1.60E 02 -1.20E 02 -8.00E 01 -4.00E 01 0.00E 00 4.00E 01 8.00E 01 1.20E 02 1.60E 02 2.00E 02  
 -2.000E 00 ..... \* I  
 -1.750E 00 ..... \* I  
 -1.500E 00 ..... \* I  
 -1.250E 00 ..... \* I  
 -1.000E 00 ..... \* I  
 -7.500E-01 ..... \* I  
 0.000E 00 ..... \* I  
 -5.000E-01 ..... \* I  
 -2.500E-01 ..... \* I  
 -9.5367E-07 ..... \* I  
 2.500E-01 ..... \* I  
 5.000E-01 ..... \* I  
 7.500E-01 ..... \* I  
 1.000E 00 ..... \* I

## IVC USE OF COMPENSATION

When the desired behavior of a control system cannot be obtained by the gain adjustment alone, compensation techniques must be used. Compensation means to improve the system performance by reshaping the open-loop transfer function characteristic of the system. The compensation of a control system can often be accomplished either by an element in series with other components as shown in Fig. 4.4a or by an element in parallel with one or more components

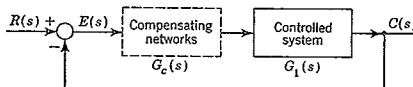


Fig. 4.4a Cascade compensation

to form a subsidiary loop, as shown in Fig. 4.4b. The former arrangement is referred to as cascade or series compensation and the latter is called feedback or minor-loop compensation. A compensator or compensating device can

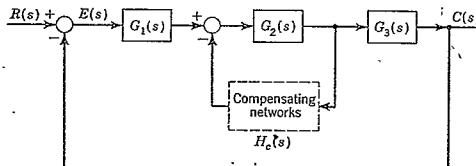


Fig. 4.4b Feedback compensation

stabilize a system which is unstable for all values of gain; it can improve both the transient and the steady-state performance of a system; and it can reduce the system error. Compensators are classified according to their operating characteristics into phase-lead (differentiating) type, phase-lag (integrating) type, and lag-lead (integro-differentiating) type. An example of the latter was given at the end of Chapter 3. The phase-lead type of compensator is generally used to modify the high-frequency portion of the open-loop transfer function plot and to improve the transient behavior of the system,

whereas the phase-lag type of compensator is often used to alter the low-frequency portion of the open-loop plot and to improve the steady-state performance of the system. The lag-lead compensation gives results intermediate between these two extremes. By proper adjustment of time constants, considerable flexibility of control system characteristics may be obtained.

The choice of a method of compensation generally depends upon the specific system involved, the available components, cost considerations and the designer's experience and judgment.

The most commonly used configuration for compensation is shown in Fig. 4.4a.<sup>4a</sup> The procedure to compute  $G_c(s)$  is to first compute the open loop transfer function and then compute  $G_c(s)$ . There are two disadvantages in using this configuration. First if the overall open loop transfer function  $G_f(s)$  is not properly chosen, the compensator computer in this manner may not be realizable as a RC network. Second, it generally requires pole-zero cancellation. In system theory terminology, it means that some poles of the overall system are uncontrollable and/or unobservable. This poses a design problem in that these poles are dictated by the given plant  $G_1(s)$  and cannot be controlled by the designer.

The feedback compensation, shown in Fig. 4.4b, is sometimes superior to cascade compensation in that variation of the parameters of the system components bridged by the feedback elements of the minor loop have less effect upon system performance if the minor-loop gain is made sufficiently large and if the parameters of the feedback compensator do not vary. Although the freedom in choosing  $G_1(s)$  and  $H_c(s)$  is greatly increased, the difficulties encountered in Fig. 4.4a may still occur. Therefore the series and shunt compensation are not always applicable in practice.

Consider the configuration of compensators introduced by Chen [CH 1]. This is shown in Fig. 4.5, where  $k$  is a constant gain,  $C_1(s) = \frac{N_1(s)}{D_1(s)}$  and

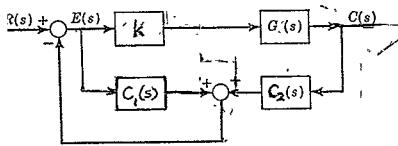


Fig. 4.5 Chen's Control System Configuration

$C_2(s) = \frac{N_2(s)}{D_2(s)}$  are proper rational functions with the degree of polynomials  $D_1(s)$ ,  $D_2(s)$  and  $N_2(s)$  equal to  $n-1$  and that of  $N_1(s) = n-2$ . This control system configuration has the overall transfer function

$$G_f(s) = \frac{kG(s)}{1 + C_1(s) + kC_2(s) G(s)} \quad (4.6)$$

$$= \frac{kND_2 D_1}{D D_1 D_2 + N_1 D_2 D + kN_2 N D_1}$$

If the denominators of  $C_1(s)$  and  $C_2(s)$  are chosen to be the same, that is  $D_1(s) = D_2(s)$ , then the last equation reduces to

$$G_f(s) = \frac{kND_1}{DD_1 + N_1 D + kN_2 N} \quad (4.7)$$

It can be shown that by using this system configuration the compensators can be always chosen to be realizable by RC networks and the cancelled poles can also be controlled by the designer [see CH 1]. It is worth noting that the complexity of compensators in the system of Fig. 4.5 is comparable to that required for the compensator of the corresponding control system based on the configuration of Fig. 4.4a.

### Examples

An example of cascade compensation was included at the end of Chapter III (refer to Figs. 3.16 and 3.17). Here we provide the Bode plots in Figs. 4.6 and 4.7 for the uncompensated as well as the compensated case respectively to facilitate comparisons.

Next we illustrate basic feedback compensation with the aid of two examples having the configuration shown in Fig. 4.8. The feedback function  $H_c$  is used to modify the characteristics of the plant  $G$ , while the cascade function  $G_f$  is provided to aid in adjusting the performance of the major loop. Often the cascade compensation  $G_f$  is a simple gain factor used to adjust the degree of stability of the system. The burden of modifying the transfer function  $G$  is placed on the feedback compensation  $H_c$ . In addition, the feedback function is usually provided with an adjustable gain factor to permit setting the degree of stability of the minor loop.

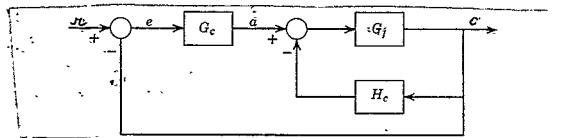


Fig. 4.8 General arrangement for feedback compensation.

The procedure for adjusting the feedback compensation can be based on the magnitude asymptotes of the minor loop. In the frequency ranges where the open-minor-loop frequency-response magnitude  $|G(j\omega) H_c(j\omega)|$  is very large, the closed-minor-loop response  $|c(j\omega)/a(j\omega)|$  behaves like the reciprocal of the feedback compensation. When the open-minor-loop response magnitude is very small, the closed-minor-loop response behaves like the plant.

The feedback compensation is used primarily to improve the dynamic behavior

of the plant in the mid-frequency range without altering the high gain of the plant at low frequencies. A high-frequency boundary will always exist since in any practical case the magnitude of  $G(j\omega) H_c(j\omega)$  will become less than unity as frequency increase.

The procedure for adjusting the feedback function  $H_c$  and the cascade gain factor [ $G_c(s) = K_c$ ] can thus be roughed out by means of asymptotic plots of the pertinent responses and then carried out in detail by means of the gain-phase or Bode plots. The asymptotic plots enable one to examine the form of the closed-minor-loop response as the feedback compensation is adjusted.

Since there are usually several parameters to adjust in the feedback compensation procedure, the process of design is one of trial and error.

NASAP PROBLEM UNCOMPENSATED PLANT

OPEN Loop BODE Plot.

RADIANS  
NONE

V1 1 2 0.83

L1 2 1 1.4

V2 1 3 1.0 IL1

R1 3 4 1

C1 4 1 1F

V3 1 5 1.0 VC1

R2 5 6 0.2

C2 6 1 1E

V4 1 7 1.0 VC2

OUTPUT

VV4/VV1/V1

NUMBER OF LOOPS PER ORDER

FREQ -2, 1.0 0.05

EXECUTE

1=

2=

3  
1

TRANSFER FUNCTION VV4/VV1/V1

H(S) = 4.150E 00\*

( 1.00E 00 )

( 0.00E 00 +5.00E 00 S +6.00E 00 S 2 +1.00E 00 S 3 )

ZERO OF TRANSFER FUNCTION

NONE

POLE OF TRANSFER FUNCTION

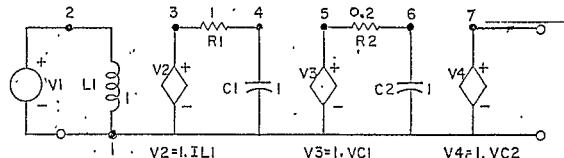
POLE REAL PART IMAG. PART

1 0.0000E 00 0.0000E 00

2 -0.10000E 01 0.0000E 00

3 -0.50000E 01 0.0000E 00

$$s(1+s)(1+0.2s)$$



Dig 1.6

Ldg (Freq)	Freq	20*Ldg (Abs(H))	Phi(H)	Abs(H)	Ldg (Abs(H))
-0.2000000E_01	0.9999998E-02	0.3638106E_02	-0.96868759E_02	0.8299545E_02	0.191905E_01
-0.1950000E_01	0.1122019E+01	0.3738094E_02	-0.9077148E_02	0.7396673E_02	0.1869047E_01
-0.1900000E_01	0.1251426E+01	0.3638080E_02	-0.9086465E_02	0.6592366E_02	0.1819040E_01
-0.1849999E_01	0.1412539E+01	0.3536063E_02	-0.9097119E_02	0.5875225E_02	0.1769032E_01
-0.1800000E_01	0.1544492E+01	0.3436039E_02	-0.9108956E_02	0.5236223E_02	0.1719020E_01
-0.1750000E_01	0.1771279E+01	0.3336010E_02	-0.9124261E_02	0.4666665E_02	0.1669005E_01
-0.1700000E_01	0.1993261E+01	0.3227971E_02	-0.9131777E_02	0.4158898E_02	0.1618986E_01
-0.1650000E_01	0.2238723E+01	0.3137926E_02	-0.9153906E_02	0.3706499E_02	0.1568963E_01
-0.1600000E_01	0.2511184E+01	0.3037716E_02	-0.912685E_02	0.3303220E_02	0.1518934E_01
-0.1550000E_01	0.2810392E+01	0.2937793E_02	-0.9193741E_02	0.2943729E_02	0.1466897E_01
-0.1500000E_01	0.3112277E+01	0.2837700E_02	-0.9217369E_02	0.2523315E_02	0.1418850E_01
-0.1450000E_01	0.3456135E+01	0.2737502E_02	-0.9243874E_02	0.2337721E_02	0.1367892E_01
-0.1400000E_01	0.3931106E+01	0.2637473E_02	-0.9273607E_02	0.2083145E_02	0.1318719E_01
-0.1350000E_01	0.4456032E+01	0.2537251E_02	-0.9306952E_02	0.1856210E_02	0.1248626E_01
-0.1300000E_01	0.5011140E+01	0.2431016E_02	-0.9324435E_02	0.1553903E_02	0.1218509E_01
-0.1250000E_01	0.5623431E+01	0.2336726E_02	-0.9338630E_02	0.1473547E_02	0.1168364E_01
-0.1200000E_01	0.6302563E+01	0.2236359E_02	-0.9434340E_02	0.1312749E_02	0.1118179E_01
-0.1150000E_01	0.7079446E+01	0.2135896E_02	-0.9480797E_02	0.1169361E_02	0.1067948E_01
-0.1100000E_01	0.7942273E+01	0.2015312E_02	-0.9545187E_02	0.1041696E_02	0.1017456E_01
-0.1050000E_01	0.8911204E+01	0.1934579E_02	-0.96111426E_02	0.9427449E_01	0.9472902E_00
-0.1000000E_01	0.9937495E+01	0.1833655E_02	-0.9646460E_02	0.8257152E_01	0.9168300E_00
-0.95000017E_00	0.1122014E+00	0.1712504E_02	-0.97676747E_02	0.7349417E_01	0.8662525E_00
-0.90000015E_00	0.1258921E+00	0.1631052E_02	-0.98381774E_02	0.6539229E_01	0.8155259E_00
-0.85000013E_00	0.1412533E+00	0.1529294E_02	-0.9965829E_02	0.5815876E_01	0.7664147E_00
-0.80000011E_00	0.1564449E+00	0.1426945E_02	-0.1008215E_03	0.5167971E_01	0.7134727E_00
-0.75000019E_00	0.1770272E+00	0.1324068E_02	-0.1021203E_03	0.4592444E_01	0.6620438E_00
-0.70000017E_00	0.1955254E+00	0.1220513E_02	-0.1035590E_03	0.4076210E_01	0.6102565E_00
-0.65000019E_00	0.2239714E+00	0.1116050E_02	-0.1048266E_03	0.3614305E_01	0.5580248E_00
-0.60000019E_00	0.2511179E+00	0.1010404E_02	-0.1057635E_03	0.3200102E_01	0.5052448E_00
-0.55000021E_00	0.2818396E+00	0.9035851E_02	-0.1048961E_03	0.2830042E_01	0.4517825E_00
-0.50000019E_00	0.3162264E+00	0.9250322E_02	-0.11111673E_03	0.2497516E_01	0.3975161E_00
-0.45000017E_00	0.34548120E+00	0.6844790E_02	-0.1135944E_03	0.2190973E_01	0.3422396E_00
-0.40000015E_00	0.3891057E+00	0.5715205E_02	-0.1162620E_03	0.1930902E_01	0.287603E_00
-0.35000023E_00	0.44668138E+00	0.4556966E_01	-0.1191747E_03	0.1689851E_01	0.2278483E_00
-0.30000021E_00	0.5011848E+00	0.3349490E_01	-0.1223343E_03	0.1473150E_01	0.1612470E_00
-0.25000019E_00	0.5623308E+00	0.2133668E_01	-0.1257677E_03	0.1276452E_01	0.1066843E_00
-0.20000017E_00	0.6302949E+00	0.1852735E_01	-0.1294423E_03	0.1104768E_01	0.4227793E_00
-0.15000025E_00	0.7079417E+00	0.14689473E_00	-0.1345515E_03	0.9474425E_00	-0.2347126E_01
-0.10000023E_00	0.7543240E+00	0.1815164E_01	-0.13747078E_03	0.8080659E_00	-0.9255320E_01
-0.50002110E+01	0.86912446E+00	-0.2233255E_01	-0.14181517E_03	0.844428E_00	-0.1646428E_00
-0.19027349E+05	0.9999755E+00	0.4759515E_01	-0.1620398E_03	0.555048E_00	-0.2399518E_00
-0.49997335E+01	0.1122011E+01	0.63707751E_01	-0.1593035E_03	0.4802439E_00	-0.3185391E_00
-0.99997522E+00	0.1258919E+01	0.80297475E_01	-0.1856711E_03	0.3766606E_00	-0.1048747E_00
-0.14999777E+00	0.1412535E+01	0.97161179E_01	-0.16407479E_03	0.5267305E_00	-0.4850100E_00
-0.19995795E+00	0.1584488E+01	0.1148955E_02	-0.1653272E_03	0.2663914E_00	-0.5747479E_00
-0.24999715E+00	0.17712426E+01	0.1332894E_02	-0.17022715E_03	0.2455524E_00	-0.6644468E_00
-0.29995735E+00	0.1992250E+01	0.1523321E_02	-0.1721348E_03	0.1731147E_00	-0.7616658E_00
-0.34999755E+00	0.22339708E+01	0.1720198E_02	-0.1799494E_03	0.1386069E_00	-0.8600891E_00
-0.39999777E+00	0.2511875E+01	0.1923463E_02	-0.1702342E_03	0.1921212E_00	-0.9217224E_00
-0.44999769E+00	0.2818363E+01	0.2133156E_02	-0.1701267E_03	0.1678706E_00	-0.1066578E_01
-0.49999711E+00	0.31622297E+01	0.2349332E_02	-0.1652222E_03	0.1658840E_00	-0.1146771E_01
-0.54999735E+00	0.3594112E+01	-0.2572151E_02	-0.1605793E_03	0.15175151E_00	-0.1266076E_01
-0.59999755E+00	0.39381049E+01	0.28106645E_02	-0.1522732E_03	0.1374529E_00	-0.1408828E_01
-0.64999765E+00	0.44668081E+01	0.30581810E_02	-0.1508426E_03	0.13022315E_00	-0.1518941E_01
-0.69999765E+00	0.5111137E+01	0.32100432E_02	-0.14621261E_03	0.12288632E_00	-0.1610422E_01
-0.74999715E+00	0.56233276E+01	0.35304694E_02	-0.14172252E_03	0.117171115E_00	-0.1765202E_01
-0.79999711E+00	0.61717125E+01	0.37474011E_02	-0.13429728E_03	0.1278926L_00	-0.1892156E_01
-0.84999766E+00	0.7075402E+01	-0.4044820E_02	-0.1332728E_03	0.9460025E_02	-0.2024107E_01
-0.89999685E+00	0.7943232E+01	0.43156111E_02	-0.1293648E_03	0.6954863E_02	-0.2157836E_01
-0.94999695E+00	0.8912447E+01	-0.4580158E_02	-0.1266950E_03	0.5080663E_02	-0.2294079E_01
-0.99999715E+00	0.99999494E+01	-0.48505117E_02	-0.1222759E_03	0.30693514E_02	-0.2429259L_01

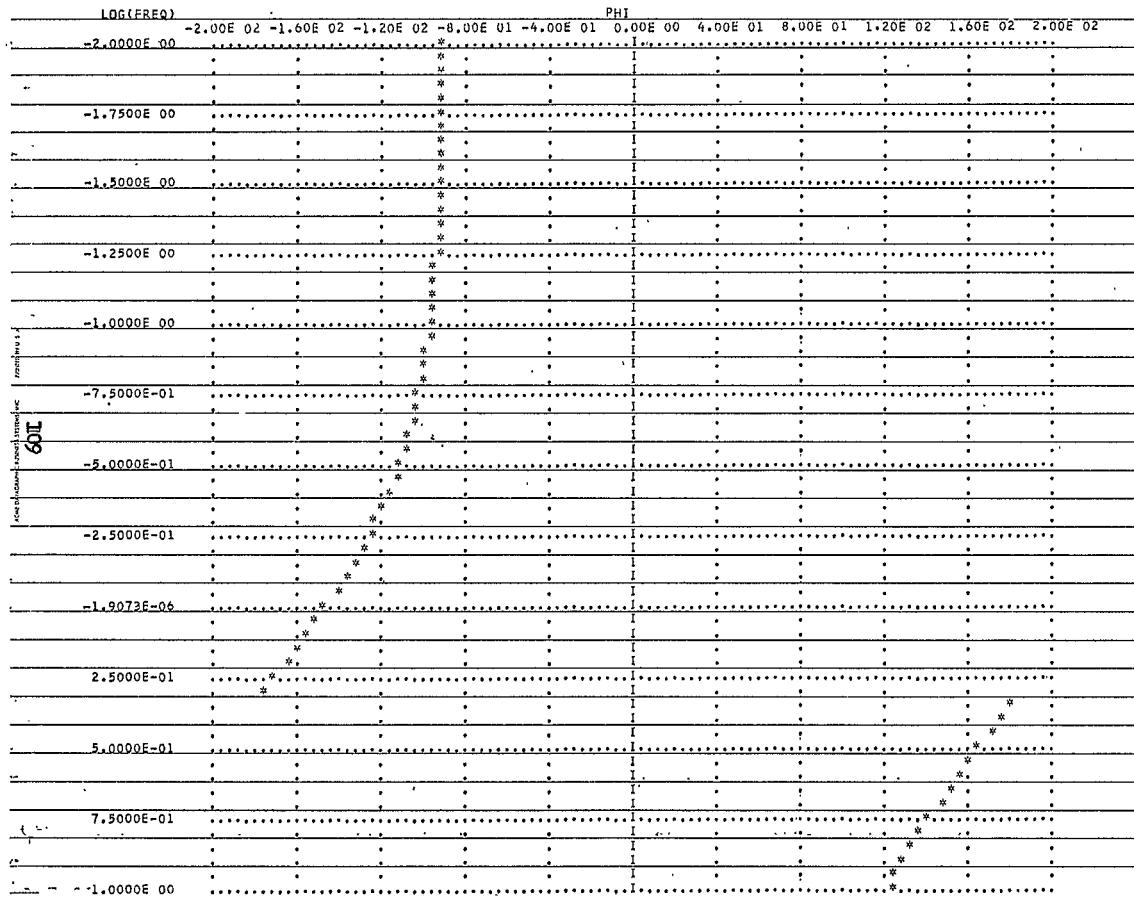
LOG(FREQ)

20.\*LOG(ABS(H))

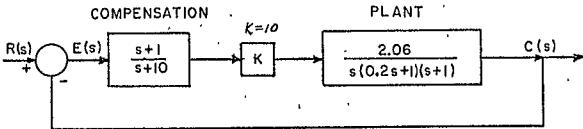
-2.000E-01 -1.750E-01 -1.500E-01 -1.250E-01 -1.000E-01 -7.500E-02 -5.000E-02 -2.500E-02 -1.5073E-02 2.5000E-01 5.0000E-01 7.5000E-01 1.0000E+00

RE(z)

IM(z)



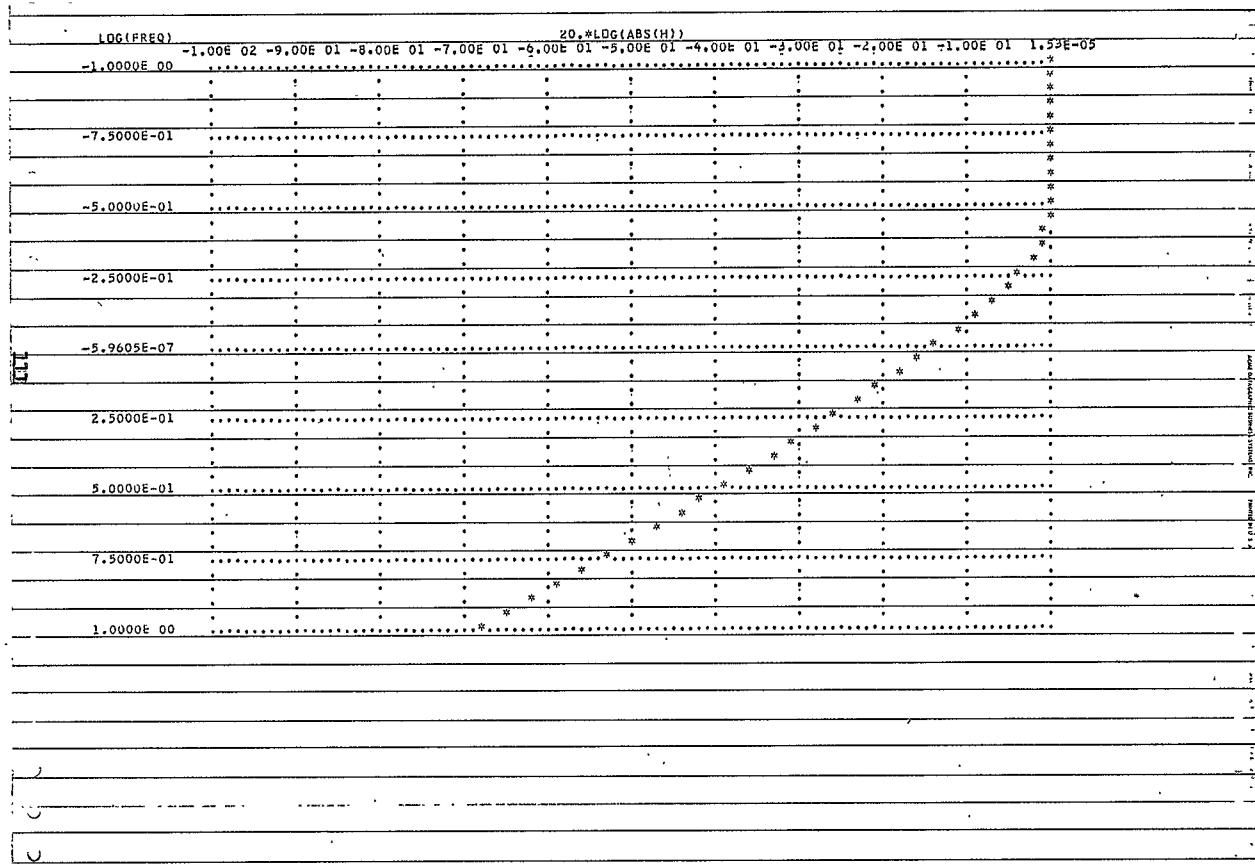
L0G(FREQ)	FREQ	20.*L0G(ABS(H))	PHI(H)	ABS(H)	L0G(ABS(H))
-0.9999996E+00	0.1000001E+00	0.9777428E-01	-0.1781549E+02	0.1010157E+01	0.4388715E-02
-0.999997E+00	0.1122019E+00	-0.2008229E+02	0.1012525E+02	0.546655826E-02	
-0.999997E+00	0.1258926E+00	-0.2266541E+02	0.1015344E+01	0.6613061E-02	
-0.8499990E+00	0.1412233E+00	-0.1603050E+02	-0.2505164E+02	0.1018627E+01	0.615253E-02
-0.7999998E+00	0.1584494E+00	-0.1919001E+02	-0.2900237E+02	0.1022232E+01	0.5990007E-02
-0.7499999E+00	0.1778240E+00	0.2252238E+02	-0.3204808E+02	0.1026269E+01	0.1126119E-01
-0.6999995E+00	0.1995263E+00	0.2571664E+02	-0.3725061E+02	0.1030050E+01	0.1285842E-01
-0.6500000E+00	0.2228721E+00	0.2808114E+02	-0.4287262E+02	0.1032858E+01	0.1404057E-01
-0.6000000E+00	0.2511886E+00	0.2836820E+02	-0.4889008E+02	0.1033119E+01	0.1418101E-01
-0.5500001E+00	0.2816348E+00	0.2441449E+02	-0.5604611E+02	0.1028507E+01	0.1220735E-01
-0.5000001E+00	0.3162277E+00	0.2127171E+02	-0.6464694E+02	0.1014750E+01	0.63508854E-02
-0.4500002E+00	0.3484813E+00	-0.1194394E+03	-0.7472162E+02	0.9863431E+00	-0.5971968E-02
-0.4000002E+00	0.3818070E+00	-0.8529583E+02	0.9371495E+00	-0.2819110E+01	
-0.3500003E+00	0.4466833E+00	-0.1277784E+01	-0.9733092E+02	0.8631923E+00	-0.6307242E+01
-0.3000003E+00	0.5011868E+00	-0.2312713E+01	-0.1099373E+03	0.7662337E+00	-0.1156387E+00
-0.2500004E+00	0.5623409E+00	-0.3675838E+01	-0.1242468E+03	0.6549464E+00	-0.1837942E+00
-0.2000004E+00	0.6309266E+00	-0.5328114E+01	-0.1343181E+03	0.5414953E+00	-0.2664052E+00
-0.1500005E+00	0.7079450E+00	-0.7204906E+01	-0.1452428E+03	0.4362663E+00	-0.3602483E+00
-0.1000005E+00	0.7943273E+00	-0.9242250E+01	-0.1514548E+03	0.3450542E+00	-0.4621125E+00
-0.5000055E-01	0.8912498E+00	-0.1139052E+01	-0.1641129E+03	0.2696446E+00	-0.5695277E+00
-0.2965464E-06	0.9999986E+00	-0.1802E+00	-0.1722763E+03	0.2084963E+00	-0.6809011E+00
0.4999646E-01	0.1122015E+01	-0.1590626E+02	-0.1777832E+03	0.1601979E+01	-0.7953428E+00
0.9999818E-01	0.1258922E+01	-0.1828182E+02	-0.1732430E+03	0.1223359E+01	-0.9124460E+00
0.1499790E+00	0.1412235E+01	-0.2064117E+02	-0.1657096E+03	0.9288013E+01	-0.1032076E+01
0.1999920E+00	0.1584490E+01	-0.2308481E+02	-0.1605530E+03	0.7010663E+01	-0.1154241E+01
0.2499983E+00	0.1778273E+01	-0.2553796E+02	-0.1547348E+03	0.5260354E+01	-0.1278987E+01
0.2999986E+00	0.1995256E+01	-0.2812703E+02	-0.1492349E+03	0.3923236E+01	-0.1406352E+01
0.3499983E+00	0.2238715E+01	-0.3072526E+02	-0.1470467E+03	0.2903516E+01	-0.1530328E+01
0.3999990E+00	0.2511881E+01	-0.3337682E+02	-0.1591703E+03	0.2143671E+01	-0.1668841E+01
0.4499983E+00	0.2818372E+01	-0.3607511E+02	-0.1346091E+03	0.1571231E+01	-0.1803760E+01
0.4999985E+00	0.3162267E+01	-0.3881822E+02	-0.130652E+03	0.1145746E+01	-0.1940911E+01
0.5499986E+00	0.3548122E+01	-0.4160139E+02	-0.12547381E+03	0.8316088E+01	-0.2080808E+01
0.5999986E+00	0.3928106E+01	-0.44420275E+02	-0.1282283E+03	0.6011199E+02	-0.2221038E+01
0.6499981E+00	0.4466616E+01	-0.49727100E+02	-0.1192136E+03	0.4329611E+02	-0.2363550E+01
0.6999983E+00	0.5011452E+01	-0.5014796E+02	-0.1164995E+03	0.3108854E+02	-0.2507399E+01
0.7499985E+00	0.5623349E+01	-0.5304749E+02	-0.1117542E+03	0.2226512E+02	-0.2652374E+01
0.7999986E+00	0.6309254E+01	-0.5596554E+02	-0.1112729E+03	0.1591134E+02	-0.2798292E+01
0.8499979E+00	0.7079423E+01	-0.5889972E+02	-0.1090334E+03	0.1135043E+02	-0.2947986E+01
0.8999981E+00	0.7943247E+01	-0.6184464E+02	-0.1070167E+03	0.8084897E+03	-0.3092324E+01
0.9499983E+00	0.8912474E+01	-0.6480307E+02	-0.1052044E+03	0.5751195E+03	-0.32401184E+01
0.9999985E+00	0.9999965E+01	-0.6776936E+02	-0.1035783E+03	0.4088187E+03	-0.3388469E+01

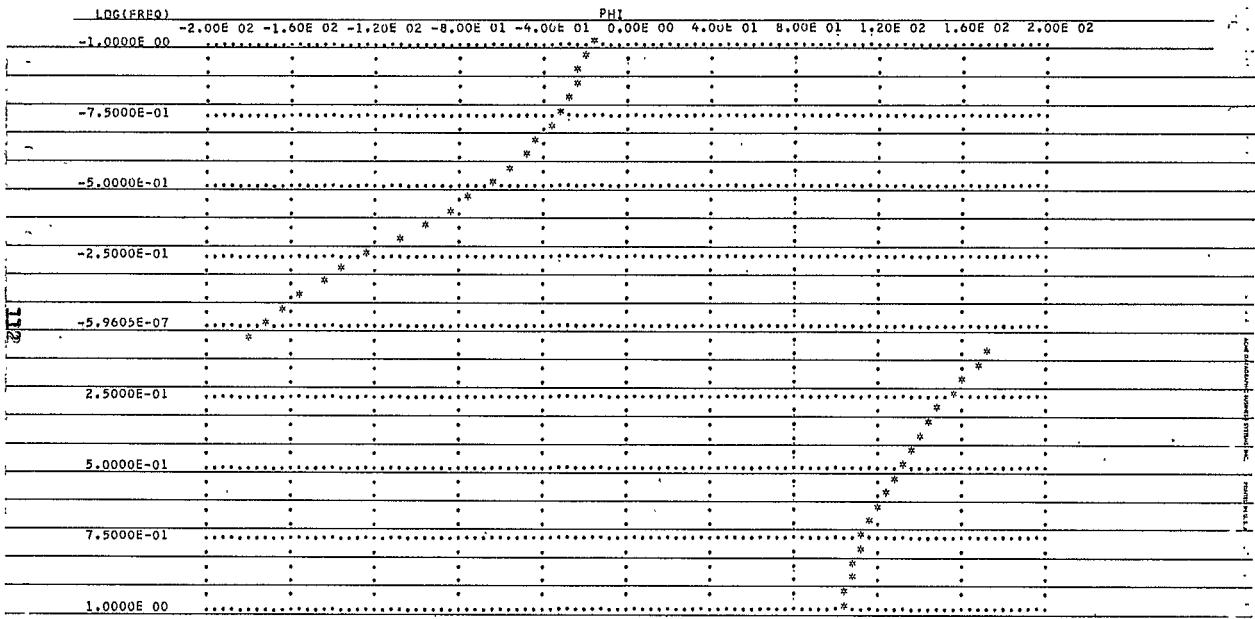


For the above unity feedback control system with lead compensation, get Bode plot.

Dig #7

noch





This example, shown in Fig. 4.9 along with its NASAP circuit model, is taken from Newton Gould and Kaiser [NE 1 pp. 326-333]. One finds that the magnitude plot of the closed-loop transfer function shows no sign of resonance effects so that a reasonable closed-loop performance may be expected. The corresponding phase curve is given in Fig. 4.10.

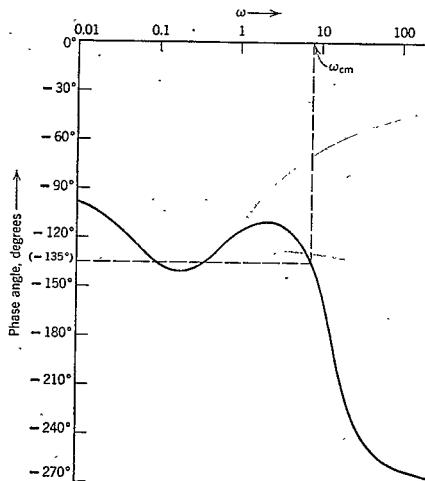


Fig. 4.10 Closed-minor-loop phase-angle response.

Using the  $45^\circ$  phase-margin criterion to adjust the cascade compensation, the magnitude crossover frequency of the major loop is found to be

$$\omega_{cm} = 7.7 \text{ rad sec}^{-1} \quad (4.8)$$

and the corresponding gain factor of the cascade compensation is

$$K = 38 \quad (4.9)$$

If this control system exhibits undesirable performance because of the lag-compensation effect in the open-major-loop response at low frequencies, some

improvement can be expected by decreasing the feedback compensation time constant  $T_c$  and by some increase in the minor-loop gain factor  $K_c$ . The degree of improvement achievable must be ascertained by further NASAP runs. Further discussion of this example is found in Chapter V including step response and error response for the control system with its gain  $K = 38$ .

NASA P. 326-333  
NEWTON-GOULD-KAISER

OUTPUT RESPONSE

RADIANS  
STEP

V8 1 13 1.  
R4 13 7 1.  
V9 1 12 38 IR4  
R5 12 11 1.  
V1 1 2 1. IR5  
R1 2 3 0 3  
C1 3 1 1F  
V2 1 4 1. V1  
R2 4 5 0 1  
C2 5 1 1F  
V3 1 6 1. VC2  
L1 6 1 1H  
V4 1 7 1. IL1  
R3 7 8 2  
C3 8 1 1F  
V5 1 9 1. VG3  
C4 9 1 1F  
V6 1 10 10 IC4  
C5 10 1 1F  
V7 1 11 1. IC5  
OUTPUT  
VV4/VVB

FREQ -1.0 2.0 0.05  
TIME -2.0  
EXECUTE

TRANSFER FUNCTION VV4/VVB

$$(-5.00E-01 + 1.00E-03 S)$$

$$H(S) = 1.267E-03 *$$

$$(-6.33E-02 + 1.28E-03 S + 2.07E-02 S^2 + 1.38E-01 S + 1.00E-00 S^4)$$

NUMBER OF LOOPS PER ORDER

1#	6
2#	5
3#	1

ZERO OF TRANSFER FUNCTION

POLE OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

POLE REAL PART IMAG. PART

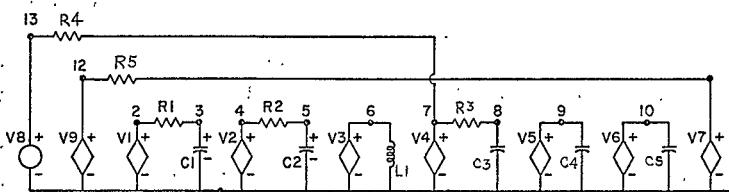
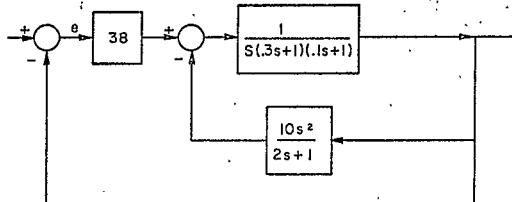
1 -0.5000E 00 0.0000E 00

1 -0.53860E 00 0.00000E 00

2 -0.28810E 01 -0.12157E 02

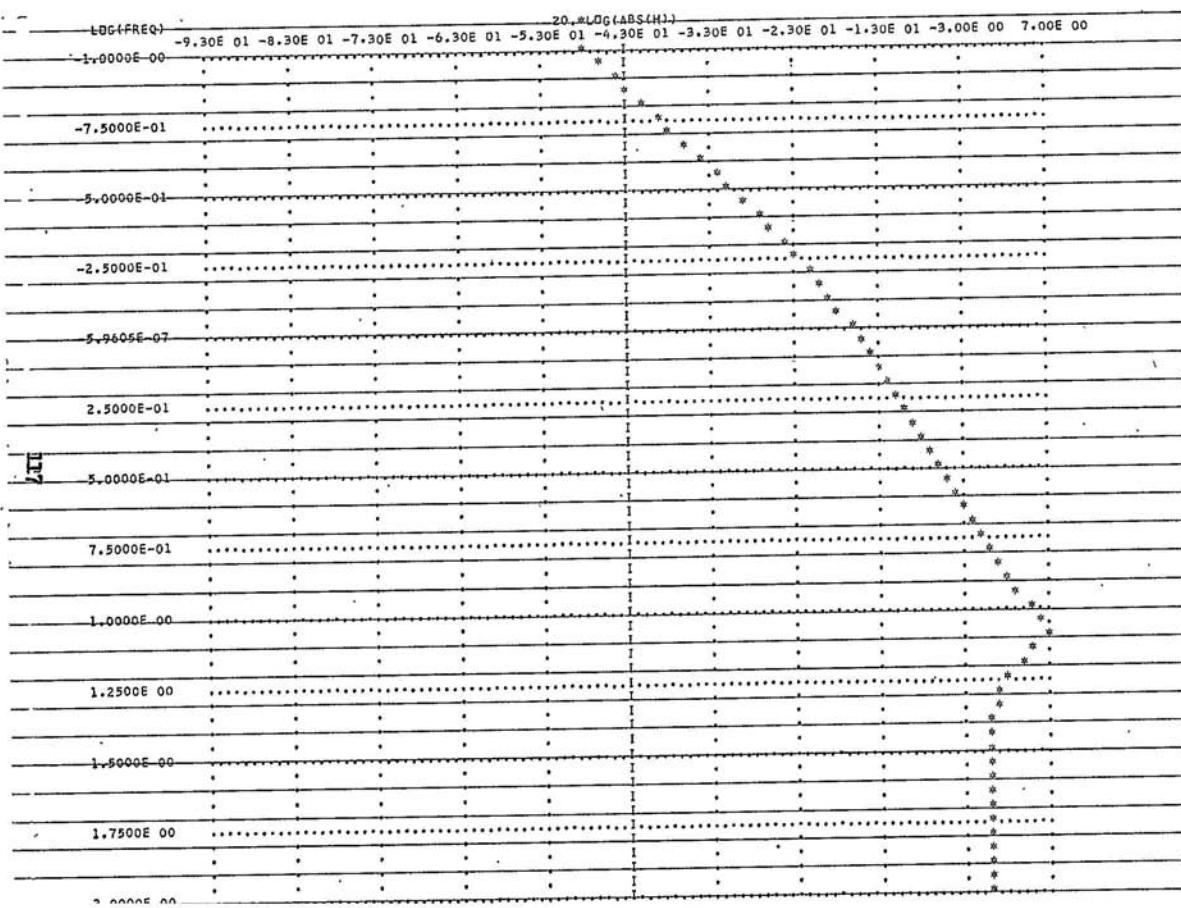
3 -0.28810E 01 -0.12157E 02

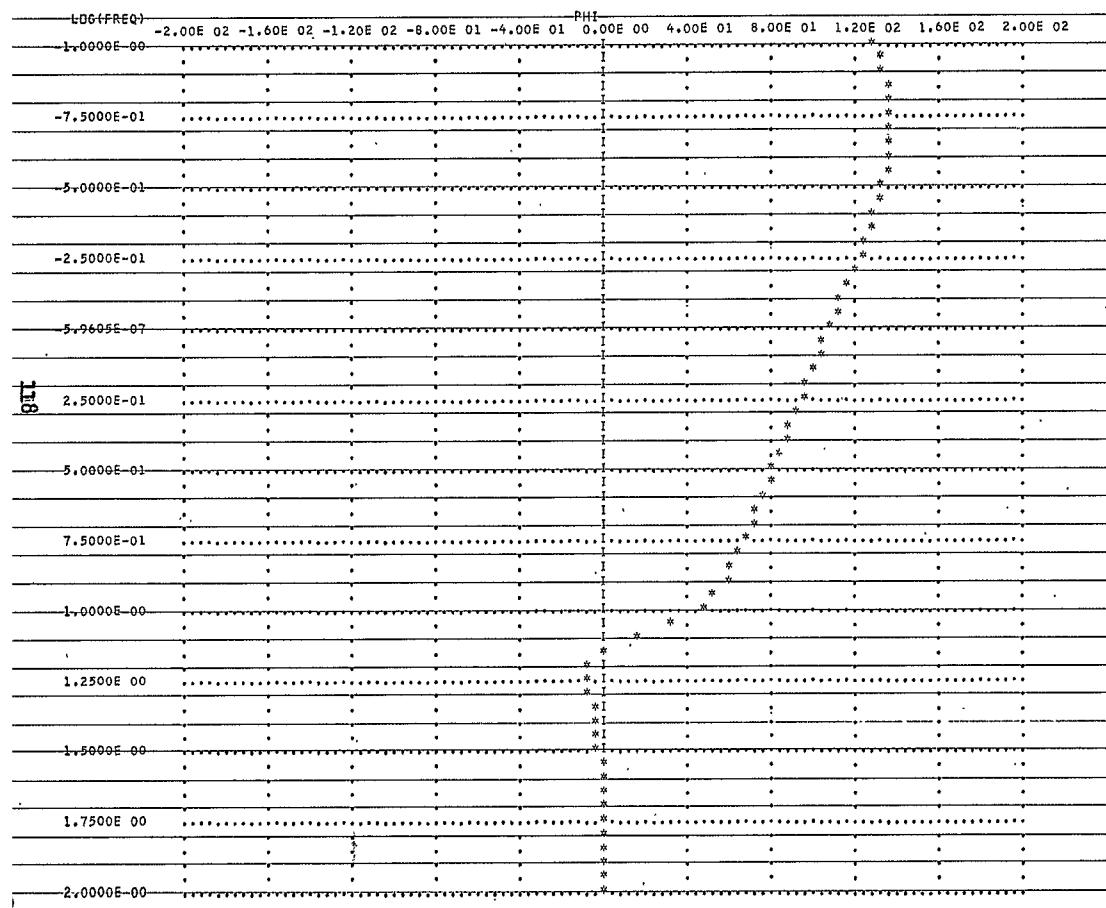
4 -0.75327E 01 0.00000E 00



2499 a

LDG(FREQ)	FREQ	20.*LDG(ABS(H))	PHI(H)	ABS(H)	LOG(ABS(H))
-0.8999996E-00	0.10000015E-00	-0.4772751E-02	0.1298571E-03	0.4107933E-02	-0.2386375E-01
-0.4999997E-00	0.11220191E-00	-0.4613454E-02	0.1317182E-03	0.4934825E-02	-0.2306727E-01
-0.8999997E-00	0.12589265E-00	-0.4449901E-02	0.1333188E-03	0.5957287E-02	-0.2224951E-01
-0.8499998E-00	0.1412538E-00	-0.4282719E-02	0.1346163E-03	0.7221695E-02	-0.2141360E-01
-0.7999995E-00	0.1584954E-00	-0.4112643E-02	0.1345759E-03	0.8783691E-02	-0.2056322E-01
-0.7499999E-00	0.1774820E-00	-0.3940440E-02	0.1361707E-03	0.1070923E-01	-0.1970241E-01
-0.6999995E-00	0.1995263E-00	-0.3767050E-02	0.1363382E-03	0.1307537E-01	-0.1883545E-01
-0.6500000E-00	0.2238721E-00	-0.3593353E-02	0.1362026E-03	0.1597067E-01	-0.1796677E-01
-0.6000006E-00	0.2511886E-00	-0.3420155E-02	0.1365290E-03	0.1949489E-01	-0.1710078E-01
-0.5500001E-00	0.2816382E-00	-0.3248387E-02	0.1346695E-03	0.2375774E-01	-0.1624194E-01
-0.5000006E-00	0.3142277E-00	-0.3078094E-02	0.1333401E-03	0.2807696E-01	-0.1539448E-01
-0.4500002E-00	0.3494813E-00	-0.2912479E-02	0.1316652E-03	0.3497519E-01	-0.1456240E-01
-0.4000001E-00	0.3981070E-00	-0.2749858E-02	0.1296771E-03	0.4217642E-01	-0.1374929E-01
-0.3500003E-00	0.4466833E-00	-0.2591649E-02	0.1274155E-03	0.5060278E-01	-0.1295825E-01
-0.3000003E-00	0.5011866E-00	-0.2438327E-02	0.1269259E-03	0.6037195E-01	-0.1219164E-01
-0.2500004E-00	0.5623409E-00	-0.2290211E-02	0.1222572E-03	0.7159674E-01	-0.1144510E-01
-0.2000004E-00	0.6309248E-00	-0.2147447E-02	0.1194584E-03	0.8438689E-01	-0.1073724E-01
-0.1500005E-00	0.7079450E-00	-0.2010020E-02	0.1165779E-03	0.9805269E-01	-0.1005011E-01
-0.1000005E-00	0.7942373E-00	-0.1877736E-02	0.1136527E-03	0.1151111E-00	-0.9388824E-00
-0.5000005E-01	0.8912498E-00	-0.1705932E-02	0.1107315E-03	0.1332917E-00	-0.8751966E-00
-0.5956464E-00	0.9999886E-00	-0.1627575E-02	0.1078286E-03	0.1535427E-00	-0.8137707E-00
-0.4999864E-01	0.11220191E-01	-0.1508801E-02	0.1049658E-03	0.1760352E-00	-0.7544040E-00
-0.0999382E-01	0.1256522E-02	-0.1332765E-02	0.1012153E-03	0.2009673E-00	-0.6958744E-00
0.1499990E-01	0.1412393E-01	-0.1281910E-02	0.9993941E-03	0.2285621E-00	-0.6409953E-00
0.1999992E-00	0.1504890E-01	-0.1173123E-02	0.9668375E-03	0.2590677E-00	-0.5858646E-00
0.2499985E-00	0.1778273E-01	-0.1066996E-02	0.9401411E-03	0.2927532E-00	-0.5334982E-00
0.2999986E-00	0.1995256E-01	-0.9631213E-02	0.9137375E-03	0.3299083E-00	-0.4816065E-00
0.3499988E-00	0.2238715E-01	-0.8616458E-02	0.8874795E-03	0.3708318E-00	-0.4308229E-00
0.3999990E-00	0.2511881E-01	-0.7621623E-02	0.8612885E-03	0.4158328E-00	-0.3810812E-00
0.4499982E-00	0.2818372E-01	-0.6646667E-02	0.8349980E-03	0.4652288E-00	-0.3332334E-00
0.4999985E-00	0.3152267E-01	-0.5630676E-02	0.8051335E-03	0.5193572E-00	-0.2845338E-00
0.5499986E-00	0.3548122E-01	-0.4752439E-02	0.7818417E-03	0.5785994E-00	-0.2376220E-00
0.5999988E-00	0.3981061E-01	-0.3828919E-02	0.7549117E-03	0.6434414E-00	-0.1914909E-00
0.6499981E-00	0.4466816E-01	-0.2918818E-02	0.7277674E-03	0.7145935E-00	-0.1459409E-00
0.4999983E-00	0.5011152E-01	-0.2012150E-02	0.7004227E-03	0.7932178E-00	-0.1006075E-00
0.7499985E-01	0.5623393E-01	-0.1073736E-02	0.6732830E-03	0.8813163E-00	-0.5486818E-01
0.7999986E-01	0.6309554E-01	-0.15248576E-02	0.64646570E-03	0.8824424E-00	-0.7692881E-02
0.0499979E-00	0.7079423E-01	-0.5508540E-02	0.6148209E-02	0.1102917E-01	0.4254270E-01
0.0999881E-00	0.7942342E-01	-0.1964546E-02	0.5805454E-02	0.12523A04E-01	0.3822971E-01
0.0499982E-00	0.8912474E-01	-0.3244503E-01	0.5309861E-02	0.1452953E-01	0.1622521E-00
0.0999285E-00	0.9999856E-01	-0.4767366E-01	0.46633249E-02	0.1719366E-01	0.2353683E-01
0.1049997E-01	0.1122012E-02	-0.6098736E-01	0.3395124E-02	0.2018073E-01	0.3049336E-00
0.1099998E-01	0.1258918E-02	-0.6524124E-01	0.1588363E-02	0.2119368E-01	0.3262062E-00
0.1149998E-01	0.1412530E-02	-0.5379307E-01	0.16746114E-01	0.1857656E-01	0.2689654E-00
0.1199998E-01	0.1584884E-02	-0.3663709E-01	0.1726632E-01	0.1524704E-01	0.1812085E-01
0.1249997E-01	0.1778268E-02	-0.2276337E-01	0.18603992E-01	0.1299622E-01	0.1138169E-00
0.1349998E-01	0.2238708E-02	-0.8104042E-00	0.15898328E-01	0.10977793E-01	0.4052031E-01
0.1399998E-01	0.2511873E-02	-0.4825823E-00	0.14389294E-01	0.1057132E-01	0.2412912E-01
0.1449997E-01	0.2818326E-02	-0.2894176E-00	0.13173734E-01	0.10338815E-01	0.14470635E-01
0.1499998E-01	0.3162257E-02	-0.1749279E-00	0.12251619E-01	0.1020349E-01	0.8740546E-02
0.1549997E-01	0.3545112E-02	-0.1055980E-00	0.11627623E-01	0.1012348E-01	0.5329899F-02
0.1599998E-01	0.3861040E-02	-0.4536745E-01	0.11553213E-01	0.1002556E-01	0.3248235E-02
0.1649997E-01	0.4466803E-02	-0.4021325E-01	0.81180165E-00	0.10046652E-01	0.2015662E-02
0.1699997E-01	0.5011838E-02	-0.24865372E-01	0.57525133E-01	0.1002875E-01	0.1248118E-02
0.1749997E-01	0.5612374E-02	-0.15517705E-01	0.4100884E-00	0.1001789E-01	0.7758853E-03
0.1799997E-01	0.4309825E-02	-0.9678025E-02	0.2901447E-00	0.1001115E-01	0.6832012E-03
0.1849997E-01	0.7079402E-02	-0.6044535E-02	0.20272878E-00	0.1000696E-01	0.3022428F-03
0.1899997E-01	0.7043224E-02	-0.3274465E-02	0.1452414E-00	0.1000435E-01	0.1888229E-03
0.1949997E-01	0.8912447E-02	-0.23687605E-02	0.1027696E-00	0.1000273E-01	0.1184380E-03
0.1999997E-01	0.9999934E-02	-0.14802621E-02	0.7212267E-01	0.1000171E-01	0.7413107E-04





The configuration of the second multiloop example is also similar to that shown in Fig. 4.8. In this case however the  $G_c$  block is simply a direct connection as shown in Fig. 4.11. Figures 4.11a through 4.11f include the NASA P printout of the Bode plots and the step response.

NASAP PROBLEM MULTIPLE FEEDBACK SYSTEM

```

-- V1 1 2 1. IR3
-- R1 2 3 0.3
-- C1 3 1 1.F
-- V2 1 4 1.0 VC1
-- R2:4 5 0.8
-- C2 5 1 1.F
-- V3 1 6 1.0 VC2
-- L1 6 1 1H
-- V4 1 7 0.042 IL1
-- C3 7 1 1F
-- V5 1 8 .30. IC3
-- V6 1 9 95.7 IR4
-- R3 9 8 1.0
-- V7 1 10 1.
-- R4 10 7.1.
-- OUTPUT
-- VV4/VV7
-- FREQ -1.0 1.0 0.05
-- TIME 5.0
-- EXECUTE

```

NUMBER\_OF\_LOOPS\_PER\_ORDER

1= 5  
2= 1

TRANSFER\_EUNCTION VV4/VV7

( 1.00E 00 )

H(S)= 1.675E 01\*

{ -1.67E -01 +9.42E -00 S +4.58E -00 S +1.00E -00 S }

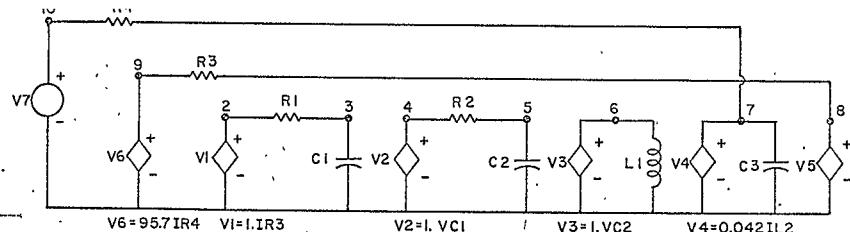
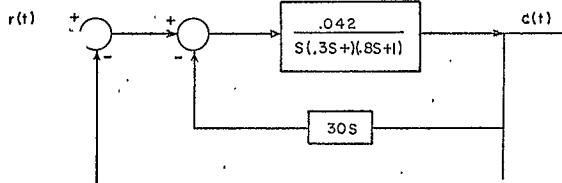
ZERO\_OF\_TRANSFER\_EUNCTION

NONE

POLE\_OF\_TRANSFER\_FUNCTION

POLE REAL PART IMAG. PART

1	-0.65674E 00	0.21657E 01	{	.658 ± j 2.766
2	-0.65474E 00	-0.21657E 01		
3	-0.32699E 01	-0.44516E -11		-3.27



24/11

LUG(FREQ)	FREQ	20.*LOG(ABS(H))	PHI(H)	ABS(H)	LOG(ABS(H))
-0.999996E-00	0.1000001E-00	0.4088327E-00	-0.2078052E-02	0.1048194E-01	0.2044164E-01
-0.999996E-00	0.1122020E-00	-0.5185689E-00	-0.2348901E-02	-0.1661521E-01	-0.2592842E-01
-0.999996E-00	0.1484874E-00	0.6581968E-00	-0.2661046E-02	0.1049977E-01	0.3293366E-01
-0.999997E-00	0.1411533E-00	0.6484464E-00	-0.3030833E-02	0.1013131E-01	0.3470793E-01
-0.999997E-00	0.1544805E-00	0.1549105E-01	0.2220355E-02	0.1130946E-01	0.5344265E-01
-0.999997E-00	0.1778282E-00	0.1364452E-01	0.3961685E-02	-0.1170099E-01	0.6622284E-01
-0.999997E-00	0.1995266E-00	0.1741468E-01	-0.4586024E-02	0.1222036E-01	0.8708405E-01
-0.999997E-00	0.2238724E-00	0.2214771E-01	0.3537238E-02	-0.1290442E-01	0.1107386E-00
-0.999997E-00	0.2411979E-00	0.2771471E-01	-0.6400209E-02	0.1371716E-01	0.1389706E-00
-0.999998E-00	0.2818337E-00	0.3361508E-01	0.1419480E-02	-0.1472569E-01	0.1680754E-00
-0.999998E-00	0.3162283E-00	0.3691287E-01	-0.5706577E-02	0.1529552E-01	0.1845644E-00
-0.999998E-00	0.3548137E-00	0.3186789E-01	0.1213195E-02	-0.1443245E-01	0.1593460E-00
-0.999999E-00	0.3911074E-00	0.1468776E-01	-0.1464752E-02	0.1170278E-01	0.6824980E-01
-0.999998E-00	0.4466839E-00	0.1468303E-01	0.1674146E-02	-0.8444711E-00	-0.7341516E-01
-0.999998E-00	0.5011875E-00	-0.4692690E-01	0.1768610E-03	0.5825933E-00	-0.2334634E-00
-0.999998E-00	0.56523416E-00	-0.7964831E-01	0.1650078E-03	-0.3997224E-00	-0.3982416E-00
-0.999998E-00	0.6309577E-00	-0.1118003E-01	0.1556891E-03	0.2760568E-00	-0.5590015E-00
-0.999999E-00	0.70494961E-00	0.1432597E-02	0.1408431E-03	-0.1921771E-00	-0.7162983E-00
-0.9999985E-01	0.7943246E-00	-0.1741710E-02	0.1415611E-03	0.1346525E-00	-0.8707854E-00
-0.9999989E-01	0.8802512E-01	-0.2046562E-02	0.1359417E-03	-0.9478039E-01	-0.1023281E-01
-0.999993E-06	0.1000000E-01	-0.7148933E-02	0.1310002E-03	0.6691647E-01	-0.1174467E-01
-0.9999959E-01	0.1122017E-01	0.2649577E-02	0.1266174E-03	-0.4733269E-01	-0.1324839E-01
-0.9999979E-01	0.1258924E-01	-0.2994948E-02	0.1227107E-03	0.3351675E-01	-0.1474744E-01
-0.1500000E-00	0.1412538E-01	-0.3248805E-02	0.1192196E-03	-0.2374632E-01	-0.1624403E-01
-0.1999992E-00	0.1584849E-01	-0.3547902E-02	0.1160962E-03	0.1682860E-01	-0.1773952E-01
-0.2499994E-00	0.1778276E-01	-0.3846954E-02	0.1133009E-03	-0.1192675E-01	-0.1923477E-01
-0.2999996E-00	0.1995260E-01	-0.4146046E-02	0.1107996E-03	0.8452315E-02	-0.2073024E-01
-0.3499998E-00	0.2238720E-01	-0.4445227E-02	0.1085622E-03	0.5989432E-02	-0.2222614E-01
-0.4000000E-00	0.2511186E-01	-0.4744510E-02	0.1065618E-03	0.4243694E-02	-0.2372255E-01
-0.4499992E-00	0.2818375E-01	-0.5043898E-02	0.1047741E-03	-0.3006455E-02	-0.2521945E-01
-0.4999994E-00	0.3162272E-01	-0.5343379E-02	0.1031771E-03	0.2129658E-02	-0.2671690F-01
-0.5499995E-00	0.35948129E-01	-0.5642295E-02	0.1017511E-03	-0.1508416E-02	-0.2821479E-01
-0.5999996E-00	0.3981067E-01	-0.5942609E-02	0.1004783E-03	0.1068303E-02	-0.2971305E-01
-0.6500000E-00	0.4466831E-01	-0.6242328E-02	0.9934241E-02	0.7565462E-03	-0.3121164E-01
-0.6999992E-00	0.5011859E-01	-0.6542096E-02	0.9832903E-02	0.5357377E-03	-0.3271048E-01
-0.7499994E-00	0.5623340E-01	-0.6841913E-02	0.9742509E-02	-0.3793531E-03	-0.3420957E-01
-0.7999996E-00	0.6309566E-01	-0.7141765E-02	0.9661897E-02	0.2686067E-03	-0.3570883E-01
-0.8499998E-00	0.7079450E-01	-0.7541646E-02	0.9590010E-02	-0.1901853E-03	-0.3720823E-01
-0.8999990E-00	0.7943260E-01	-0.7741547E-02	0.9525916E-02	0.1346562E-03	-0.3870773E-01
-0.9499992E-00	0.8912486E-01	-0.8041469E-02	0.9468772E-02	-0.9533772E-04	-0.4020735E-01

TUMOOL

TUMOOL

LOG(FREQ)	20.*LOG(ABS(H))
-1.000E 00	=9.60E-01 -8.60E-01 -7.60E-01 -6.60E-01 -5.60E-01 -4.60E-01 -3.60E-01 -2.60E-01 -1.60E-01 -6.00E-00 -4.00E-00
-7.500E-01	.....*.....
-5.000E-01	.....*
-2.500E-01	.....*
1.1921E-07	.....*
2.500E-01	.....*
5.000E-01	.....*
7.500E-01	.....*

122

LNG(FRPO) PHI  
-2.00E\_02 -1.60E\_02 -1.20E\_02 -8.00E\_01 -4.00E\_01 0.0 4.00E\_01 8.00E\_01 1.20E\_02 1.60E\_02 2.00E\_02  
-1.0000E 00 .....\*.....  
-2.5000E-01 .....\*.....  
-5.0000E-01 .....\*.....  
-7.5000E-01 .....\*.....  
-1.0000E 00 .....\*.....  
-1.2500E-01 .....\*.....  
-1.5000E-01 .....\*.....  
-1.7500E-01 .....\*.....  
-2.0000E-01 .....\*.....  
-2.2500E-01 .....\*.....  
-2.5000E-01 .....\*.....  
-2.7500E-01 .....\*.....  
-3.0000E-01 .....\*.....  
-3.2500E-01 .....\*.....  
-3.5000E-01 .....\*.....  
-3.7500E-01 .....\*.....  
-4.0000E-01 .....\*.....  
-4.2500E-01 .....\*.....  
-4.5000E-01 .....\*.....  
-4.7500E-01 .....\*.....  
-5.0000E-01 .....\*.....  
-5.2500E-01 .....\*.....  
-5.5000E-01 .....\*.....  
-5.7500E-01 .....\*.....  
-6.0000E-01 .....\*.....  
-6.2500E-01 .....\*.....  
-6.5000E-01 .....\*.....  
-6.7500E-01 .....\*.....  
-7.0000E-01 .....\*.....  
-7.2500E-01 .....\*.....  
-7.5000E-01 .....\*.....  
-7.7500E-01 .....\*.....  
-8.0000E-01 .....\*.....  
-8.2500E-01 .....\*.....  
-8.5000E-01 .....\*.....  
-8.7500E-01 .....\*.....  
-9.0000E-01 .....\*.....  
-9.2500E-01 .....\*.....  
-9.5000E-01 .....\*.....  
-9.7500E-01 .....\*.....  
-1.0000E 00 .....\*.....  
123

STEP RESPONSE FUNCTION

E(t) =

(-0.2777E 00 J 0.4199E 00 ) E  
(-0.6567E 00 J-0.2166E 01 ) T  
(-0.2777E 00 J=0.4199E 00 ) E  
(-0.3270E 01 J=0.4452E 11 ) T  
(-0.4446E 00 J 0.6053E-12 ) E  
( 0.0 J 0.0 ) T  
1.000E .01 J=0.1361E-11 ) E

$\pi(t) = \text{unit step}$

$$\lim_{t \rightarrow \infty} c(t) = 1.0 \quad \therefore \text{loss} = 0$$

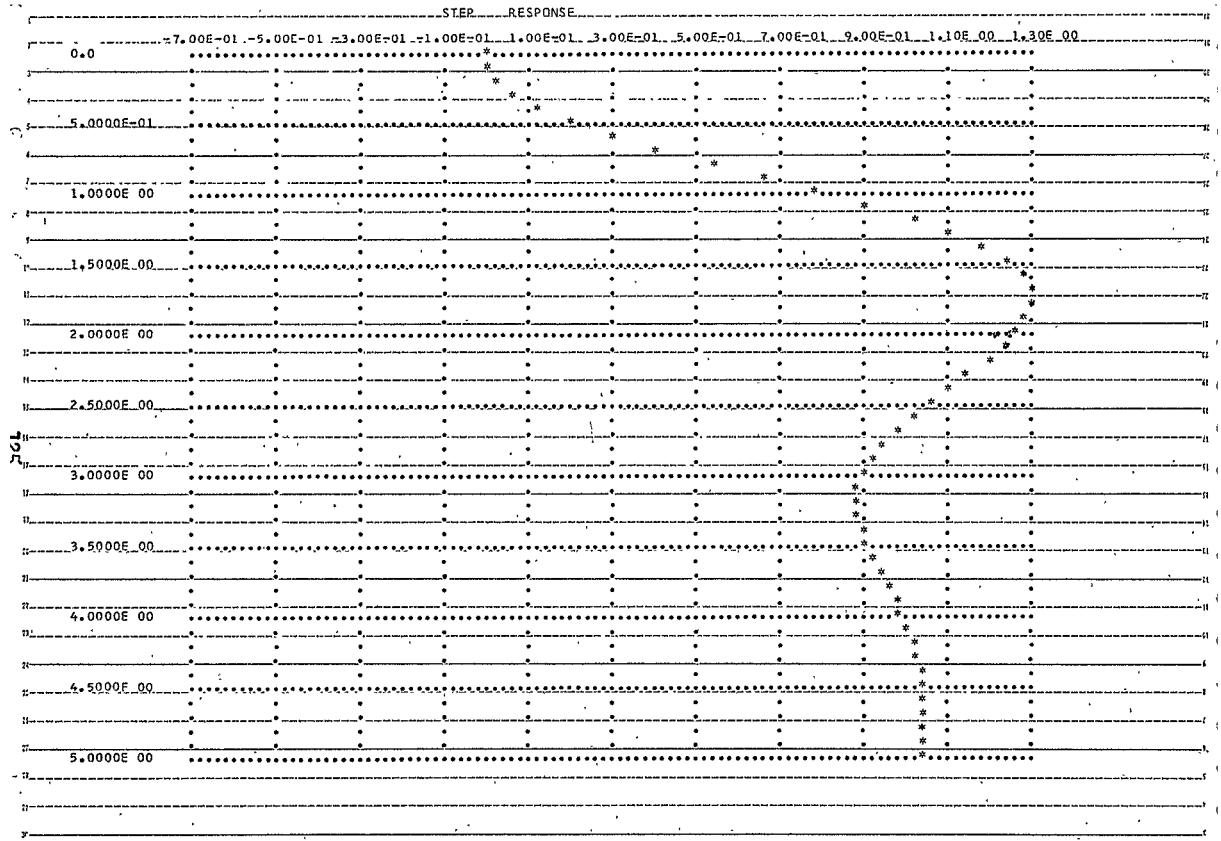
12k

STEP RESPONSE

TIME

0.0 0.0  
0.1000E 00 0.24863482E-02  
0.2000E 00 0.17693222E-01  
0.3000E 00 0.57990913E-01  
0.4000E 00 0.11118782E 00  
0.5000E 00 0.19170851E 00  
0.6000E 00 0.29158024E 00  
0.7000E 00 0.40626097E 00  
0.8000E 00 0.53033084E 00  
0.9000E 00 0.658063129E 00  
0.1000E 01 0.78388274E 00  
0.1100E 01 0.90271705E 00  
0.1200E 01 0.10102568E 01  
0.1300E 01 0.11031193E 01  
0.1400E 01 0.11789398E 01  
0.1500E 01 0.12363815E 01  
0.1600E 01 0.12750998E 01  
0.1700E 01 0.12956476E 01  
0.1800E 01 0.12993524E 01 *peak of overshoot*  
0.1900E 01 0.12881603E 01  
0.2000E 01 0.12644787E 01  
0.2100E 01 0.12310028E 01  
0.2200E 01 0.11905565E 01 *30%*  
0.2300E 01 0.11459455E 01  
0.2400E 01 0.10998249E 01  
0.2500E 01 0.10545912E 01  
0.2600E 01 0.10123026E 01  
0.2700E 01 0.97465213E 01  
0.2800E 01 0.94278491E 00  
0.2900E 01 0.91759592E 00  
0.3000E 01 0.89944178E 00  
0.3100E 01 0.88337527E 00  
0.3200E 01 0.88391238E 00 *valley of undershoot*  
0.3300E 01 0.88558934E 00  
0.3400E 01 0.89252648Z 00  
0.3500E 01 0.90376583E 00  
0.3600E 01 0.91811368E 00  
0.3700E 01 0.93475419E 00  
0.3800E 01 0.95240295F 00  
0.3900E 01 0.97015329E 00  
0.4000E 01 0.98714637C 00  
0.4100E 01 0.10026409E 01  
0.4200E 01 0.10181370E 01  
0.4300E 01 0.10271064E 01  
0.4400E 01 0.10355749E 01  
0.4500E 01 0.10412312E 01  
0.4600E 01 0.10442200E 01  
0.4700E 01 0.10447283E 01  
0.4800E 01 0.10430355E 01  
0.4900E 01 0.10329454E 01  
0.5000E 01 0.10345020E 01

A04122



CHAPTER, V

CONTROL SYSTEMS ANALYSIS IN THE TIME DOMAIN

VA INPUT SIGNALS FOR TIME RESPONSE

The impulse function is the basic tool for analysis and synthesis of linear systems. However specifically for the study of linear control systems the unit step function response is the most widely used with the unit impulse and the unit ramp functions also commonly used as test inputs.

We note in passing that these three functions are related to each other by one or more integrations or differentiations. For example the unit ramp as a function of time is the integral of the unit step function. On the other hand the unit impulse may be regarded as the derivative of the unit step function (this concept is adequate for the purposes of this manual even though this derivative does not exist in the sense of calculus).

One of the first steps in adapting NASAP to control system design was to investigate the feasibility of incorporating additional input functions. This is discussed next.

VB. ADDITIONAL INPUTS FOR CONTROL APPLICATIONS

Although the NASAP 69/I package provides the time response of the output only for an impulse excitation, the time response subroutine INV of NASAP can easily be extended to the other basic input signals often used in control theory. The necessary additions to NASAP incorporate the step and ramp excitations are given in Appendix A.

The algorithm used by NASAP to determine the residues of the poles of the transfer function assumes that the poles are simple. This poses no problem since the root-finding subroutine of NASAP will not find double or higher roots but will locate a number of simple roots in the neighborhood of the actual higher order root location. Thus for an actual double root on the negative  $\sigma$ -axis, say at

$$s_{1,2} = -a,$$

the root finding algorithm of NASAP will indicate complex root pair at

$$s_{1,2} = a \pm jb$$

where the imaginary parts of  $s_{1,2}$  are extremely small in comparison to the real part. Thus while the analytic time response for a system with higher order poles will not be correct, strictly speaking, the table and plot of discrete time values will be sufficiently accurate for practical purposes.

Given a rational transfer function

$$H(s) = \frac{C(s)}{R(s)} \quad (5.1)$$

and the Laplace transform of the excitation  $R(s)$ , the Laplace transform of the output can be expressed

$$C(s) = H(s)R(s)$$

Then, by finding the residues of the poles of  $H(s)R(s)$ , the time response of the output  $c(t)$  is readily obtained.

If the excitation is an impulse function,  $R(s) = 1$ , then the Laplace transform of the output  $C(s)$  is numerically equal to the transfer function  $H(s)$ . This is method used by NASAP to find the impulse response. Once the

transfer function and its poles have been determined, all that is necessary to find the corresponding time response of the output is to calculate the residues of the poles of the transfer function.

However, if the excitation is a unit step,  $R(s) = \frac{1}{s}$ , then the poles of the Laplace transform of the output are the poles of transfer function  $H(s)$  in addition to a pole at the origin. If  $H(s)$  has no higher order poles and no pole at the origin, then the analytic expression and the tabular values of the output will be correct. Note that the residue of the pole at the origin is the steady-state value of the output. If  $H(s)$  has higher order poles but no pole at the origin, then even though the analytic expression for the output as determined by NASAP will be somewhat in error, the tabular values will be correct. If  $H(s)$  does have a pole at the origin, then the results obtained from NASAP for a step response will be in error and will probably cause premature termination of the execution due to division by zero.

This can be seen as follows:

$$\text{Given } H(s) = \frac{K(s+a_1)(s+a_2)\dots(s+a_m)}{s(s+b_1)(s+b_2)\dots(s+b_n)} \quad (5.3)$$

Then

$$C(s) = \frac{1}{s} \frac{K(s+a_1)(s+a_2)\dots(s+a_m)}{(s+b_1)(s+b_2)\dots(s+b_n)} \quad (5.4)$$

$$\text{when } R(s) = \frac{1}{s}.$$

The residue at the pole at  $s = b_1$  can be found by

$$\begin{aligned} \text{Res}(s - b_1) &= (s + b_1) C(s) \Big|_{s=b_1} \\ &= \frac{1}{-b_1} \frac{K(a_1-b_1)(a_2-b_1)\dots(a_m-b_1)}{(-b_1)(b_2-b_1)\dots(b_n-b_1)} \end{aligned} \quad (5.5)$$

Since the algorithm used in NASAP cannot indicate double poles, it assures two distinct poles at the origin. When the algorithm proceeds to determine the residue of one of these poles at the origin, the presence of the other pole at the origin causes the denominator of (3) to go to zero and thus the

residue approaches infinity. This situation can be avoided by the addition of some elements to the original circuit to put a zero at the origin and thereby determining the ramp response of this new network. In Fig. 5.1 is given a two port network  $N$  with a source  $V$  (either an independent current or voltage source) and an element  $x$  whose voltage is the required output quantity. The transfer function of this network has a pole at the origin and it is required to find the response of the voltage across the element  $x$  to a unit step excitation.

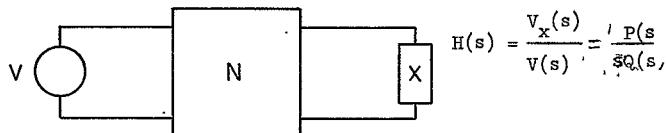


Fig. 5.1

Now consider the modification of the network in Fig. 5.1 shown in Fig. 5.2.

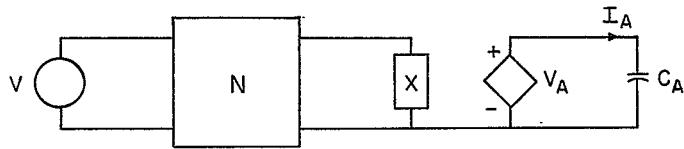


Fig. 5.2

where  $V_A$  is a dependent voltage source whose voltage equals that across the element  $x$  and  $C_A$  is a one farad capacitor. Now

$$\frac{I_A(s)}{V(s)} = \frac{sC_A V_A(s)}{V(s)} = \frac{sV_A(s)}{V(s)} \quad (5.6)$$

But  $V_A(s) = V_x(s)$

$$\text{so that } \frac{I_A(s)}{V(s)} = \frac{sV_x(s)}{V(s)} = \frac{sP(s)}{sQ(s)} = \frac{P(s)}{Q(s)} \quad (5.7)$$

Thus the pole at the origin is effectively eliminated. The response of  $i_a(t)$  to a ramp input is equivalent to the response of  $v_x(t)$  to a step input.

A similar method can be used if the output quantity is a current through an element  $x$ . In this case  $V_A$  is a dependent voltage source whose voltage equals the current through the element  $x$ .

If the excitation is a ramp,  $r(t) = t$ , then  $R(s) = \frac{1}{s^2}$ . Thus the poles of the Laplace transform of the output are the poles of  $H(s)$  in addition to the double pole at the origin. As noted earlier the algorithms of NASAP can only approximate double poles. To illustrate this assume that the double pole at the origin is approximated by a complex pole pair located on the  $j$  axis a distance  $\alpha$  from the origin. Thus

$$R(s) = \frac{1}{(s+j\alpha)(s-j\alpha)} \quad (5.8)$$

partial fraction expansion yields

$$R(s) = \frac{1}{2j\alpha} \left[ \frac{1}{s-j\alpha} - \frac{1}{s+j\alpha} \right]. \quad (5.9)$$

Taking the inverse Laplace transform yields

$$\begin{aligned} r(t) &= \frac{1}{2j\alpha} (e^{j\alpha t} - e^{-j\alpha t}) = \frac{1}{\alpha} \sin \alpha t \\ &= -j \frac{1}{2\alpha} e^{j\alpha t} + j \frac{1}{2\alpha} e^{-j\alpha t} \end{aligned} \quad (5.10)$$

However if  $\alpha t$  is small,  $\sin \alpha t$  can be approximated by  $\alpha t$ . Thus

$$i(t) \approx t \quad \text{if } \alpha t \ll 1 \quad \text{or} \quad t \ll \frac{1}{\alpha}.$$

Thus the smaller  $\alpha$  is, the larger the range of time values for which the approximation closely resembles a ramp input.

Note that the coefficients of the exponential terms in (5.10) are conjugate imaginary. However, in general, these coefficients will be complex conjugate. The significance of this is now demonstrated. Let us assume that the coefficient of  $e^{j\alpha t}$  is  $A-jB$  while that of  $e^{-j\alpha t}$  is  $A+jB$ . Thus we have

$$\begin{aligned}
 (A-jB)e^{j\alpha t} + (A+jB)e^{-j\alpha t} &= A(e^{j\alpha t} + e^{-j\alpha t}) - jB(e^{j\alpha t} - e^{-j\alpha t}) \\
 &= 2A \cos \alpha t + 2B \sin \alpha t
 \end{aligned} \tag{5.11}$$

Now let

$$2A = M \sin \theta \tag{5.12}$$

$$2B = M \cos \theta \tag{5.13}$$

Substituting (5.12) and (5.13) into (5.11) and then simplifying yields

$$(A-jB)e^{j\alpha t} + (A+jB)e^{-j\alpha t} = M \sin(\alpha t + \theta) \tag{5.14}$$

where

$$M = 2\sqrt{A^2 + B^2} \tag{5.15}$$

and

$$\theta = \tan^{-1} \frac{A}{B} \tag{5.16}$$

Now if  $A \ll B$ , then (5.15) and (5.16) can be approximated by

$$M \approx 2B \tag{5.17}$$

$$\theta \approx \frac{A}{B} \tag{5.18}$$

Since  $\theta$  and  $\alpha$  are both small, there is a range of values of  $t$  where the approximation

$$\sin(\alpha t + \theta) \approx \alpha t + \theta$$

is valid. Thus

$$(A-jB)e^{j\alpha t} + (A+jB)e^{-j\alpha t} \approx 2 \alpha B t + 2A \tag{5.19}$$

for  $A \ll B$  which represents a ramp with slope  $2\alpha B$  plus a step function of magnitude  $2A$ .

If the constant  $A$  is zero, then the above equations approximates only a ramp of slope  $2\alpha B$ . This agrees with equation (5.10). On the other hand, if  $A \gg B$ , then (5.15) and (5.16) can be approximated by

$$M \approx 2A$$

$$\theta \approx \frac{\pi}{2}$$

Thus  $\sin(\alpha t + \phi) \approx \sin(\alpha t + \frac{\pi}{2}) = \cos \alpha t$ . Since  $\alpha t$  is quite small for a wide range of time values, then

$$\cos \alpha t \approx 1$$

This results in

$$(A-jB)e^{j\alpha t} + (A+jB)e^{-j\alpha t} \approx 2A \quad (5.20)$$

where  $A \gg B$  which represents a step function of magnitude  $2A$ .

A simple example will illustrate these points. In Fig. 5.3 is the block diagram of a plant whose transfer function is known. It is required to determine the output  $c(t)$  for a ramp input of unit slope.

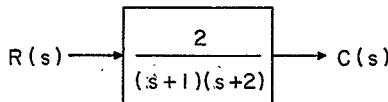


Fig. 5.3

Thus

$$C(s) = \frac{2}{s^2(s+1)(s+2)} \quad (5.21)$$

By partial fraction expansion and the inverse Laplace transform, one can obtain the exact solution

$$c(t) = t - \frac{3}{2} + 2e^{-t} - \frac{1}{2} e^{-2t} \quad (5.22)$$

In Fig. 5.4 is the equivalent circuit model for the plant given in Fig. 5.3 for computer analysis with NASAP.

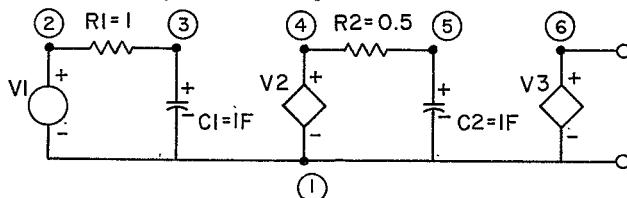


Fig. 5.4: Circuit Model for Fig. 5.3

$$V_1 = \text{input } r(t) \quad V_2 = V_{C1} \quad V_3 = V_{C2} = \text{output } c(t)$$

Desired Transfer Function  $VV_3/VV_1$ .

Since the ramp response is desired, the NASAP program will evaluate the residues of the poles of the function

$$C(s) = \frac{2}{(s+j\alpha)(s-j\alpha)(s+1)(s+2)} \quad (5.23)$$

$$\text{Res}(s = -1) = \frac{+2}{1+\alpha^2} \approx +2$$

$$\text{Res}(s = -2) = \frac{-2}{4+\alpha^2} \approx -\frac{1}{2}$$

$$\text{Res}(s = j\alpha) = \frac{-j}{\alpha(1+j\alpha)(2+j\alpha)} \approx -\frac{j}{2\alpha}$$

$$\text{Res}(s = -j\alpha) = \frac{j}{\alpha(1+j\alpha)(2+j\alpha)} \approx \frac{j}{2\alpha}$$

The computer results for the transient analysis of the circuit in Fig. 5.4 are given in Fig. 5.5. Note that for  $\alpha = 0.001$ , the coefficient A and B are -0.75 and 500 respectively. Indeed  $B > > A$  and thus by use of (5.18) we obtain

$$c(t) = t-1.5 \text{ for large } t$$

which agrees with the exact results obtained from (5.22).

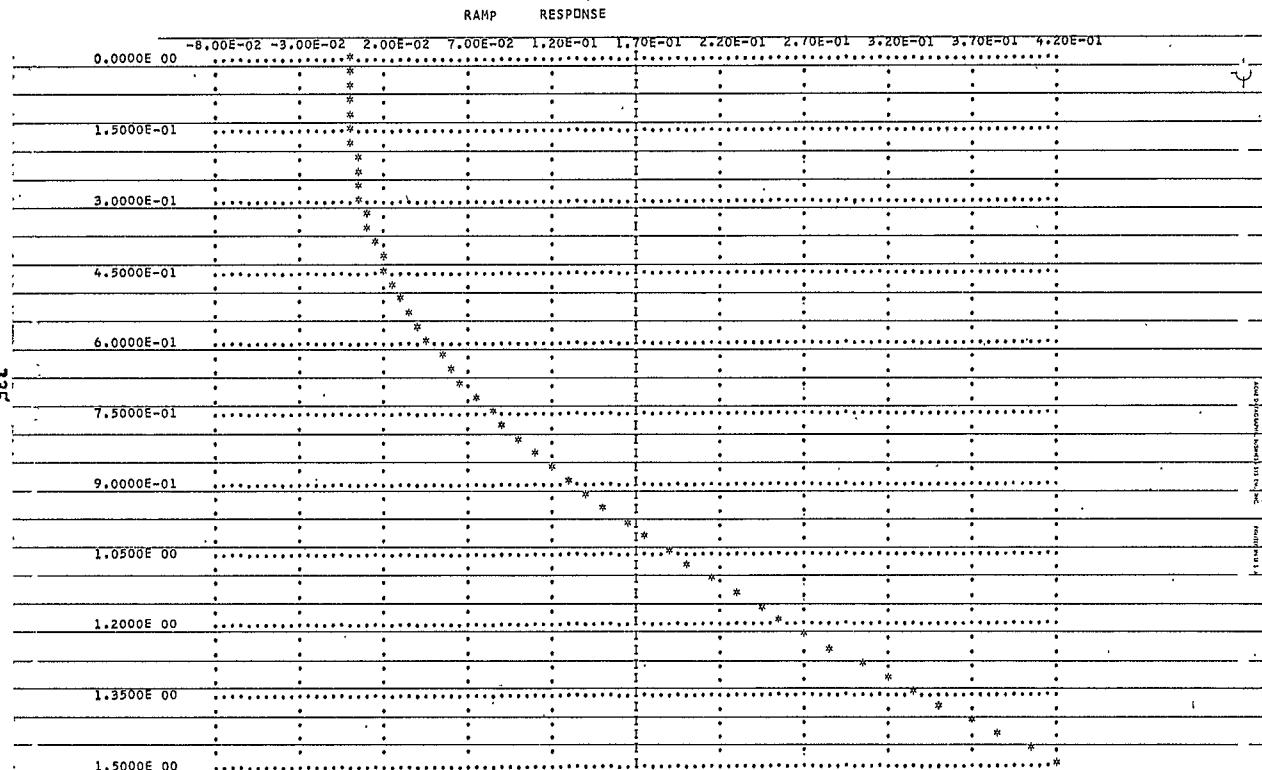
#### NASAP RAMP RESPONSE

NONE			
RAMP RESPONSE			
V1	1	2	1
R1	2	3	1
C1	3	1	1F
V2	1	4	1 VC1
R2	4	5	0.5
C2	5	1	1F
V3	1	6	1 VC2
OUTPUT			
VV3/VV1			
TIME 1.5			
EXECUTE			

## TRANSFER FUNCTION VV3/VV1

## RAMP RESPONSE

TIME	VV3/VV1
0.0000E 00	-0.22649765E-05
0.3000E-01	0.57220459E-05
0.6000E-01	0.66041946E-04
0.9000E-01	0.22459030E-03
0.1200E 00	0.52416325E-03
0.1500E 00	0.10080998E-02
0.1800E 00	0.16986132E-02
0.2100E 00	0.26420355E-02
0.2400E 00	0.38610697E-02
0.2700E 00	0.5381188E-02
0.3000E 00	0.72265863E-02
0.3300E 00	0.91836833E-02
0.3600E 00	0.11973619E-01
0.3900E 00	0.14907360E-01
0.4200E 00	0.18234968E-01
0.4500E 00	0.21957888E-01
0.4800E 00	0.26117563E-01
0.5100E 00	0.30690789E-01
0.5400E 00	0.35696208E-01
0.5700E 00	0.41138589E-01
0.6000E 00	0.47023952E-01
POLE REAL PART IMAG. PART	NUMBER OF LOOPS PER ORDER
1 -0.10000E 01 0.00000E 00	1# 3
2 -0.20000E 01 0.00000E 00	2# 1
RAMP RESPONSE FUNCTION	
F(T) =	
(-0.1000E 01 J 0.0000E 00 ) T	
( 0.2000E 01 J 0.4210E-09 ) E	
(-0.2000E 01 J 0.0000E 00 ) T	
(-0.5000E 00 J-0.5262E-10 ) E	
( 0.0000E 00 J 0.1000E-02 ) T	
(-0.7500E 00 J-0.5000E 03 ) E	
( 0.0000E 00 J-0.1000E-02 ) T	
(-0.7500E 00 J 0.5000E 03 ) E	
For $\alpha = 0.001$	
A = -0.75	
B = 500	
$B \gg A$	
thus	$2\alpha B t + 2A = t - 15$
245.56	
	12/16/69



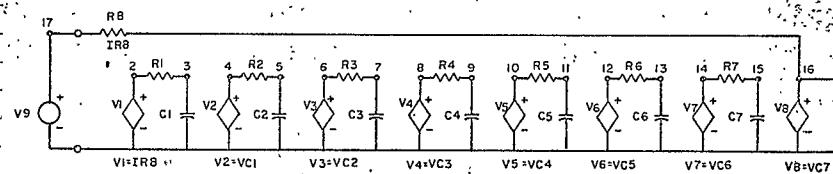
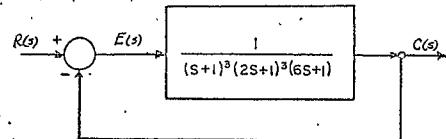
Although a few examples of NASAP printout of step response appeared earlier in this manual we include a specific example here. This step response is for the Eisenberg control problem shown previously in Fig. 4.3. The NASAP printout is shown in Fig. 5.6. We shall refer to this problem again in Chapter VII.

We are not including any examples of impulse response since all the previous references to NASAP only show such printouts.

RUNNING TIME : 48.65 SEC. on SPECTRA 70

## NASAP PROBLEM EISENBERG CONTROL SYSTEM

V9 1 17 1.  
 R8 17 10 1.  
 V1 1 2 1. IR8  
 R1 2 3 6.  
 C1 3 1 1F  
 V2 1 4 1. VC1  
 R2 4 5 2.  
 C2 5 1 1F  
 V3 1 6 1. VC2  
 R3 6 7 2.  
 C3 7 1 1F  
 V4 1 8 1. VC3  
 R4 8 9 2.  
 C4 9 1 1F  
 V5 1 10 1. VC4  
 R5 10 11 1.  
 C5 11 1 1F  
 V6 1 12 1. VC5  
 R6 12 13 1.  
 C6 13 1 1F  
 V7 1 14 1. VC6  
 R7 14 15 1.  
 C7 15 1 1F  
 V8 1 16 1. VC7  
 OUTPUT  
 VV8/VV9  
 TIME 50.  
 EXECUTE



V9 corresponds to input R(s)  
 V8 " " output C(s)  
 V1 " " error signal E(s)

## NUMBER OF LOOPS PER ORDER

1 =	9
2 =	21
3 =	35
4 =	35
5 =	21
6 =	7
7 =	1

TRANSFER FUNCTION VV8/VV9

{ 1.00E 00 }

H(S) = 2.083E-02\*

{ 4.17E-02 +3.12E-01 S +1.81E 00 S +2.44E 00 S +9.25E 00 S +9.00E 00 S +4.67E 00 S +1.00E 00 S }

## ZERO OF TRANSFER FUNCTION

NONE

## POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PAR!

1	-0.52223E 00	0.48111E 00
2	-0.52223E 00	-0.48111E 00
3	-0.72132E-01	-0.20595E 00
4	-0.72132E-01	0.20595E 00
5	-0.10790E 01	-0.38799E 00
6	-0.10790E 01	0.38799E 00
7	-0.13200E 01	-0.92015E-12

STEP	RESPONSE FUNCTION	STEP	RESPONSE
F(T) =		TIME	VV8/VV9
		0.0000E 00	0.11920929E-06
	(-0.5222E 00 J 0.4811E 00 ) T	0.1000E 01	0.26226044E-05
	(-0.7046E-01 J 0.1088E 00 ) E	0.2000E 01	0.16987324E-03
	(-0.5222E 00 J -0.4811E 00 ) T	0.3000E 01	0.16793901E-02
	(-0.7046E-01 J -0.1088E 00 ) E	0.4000E 01	0.73934197E-02
	(-0.7213E-01 J -0.2060E 00 ) T	0.5000E 01	0.21032572E-01
	(-0.9572E-01 J -0.3404E 00 ) E	0.6000E 01	0.45632005E-01
	(-0.7213E-01 J 0.2060E 00 ) T	0.7000E 01	0.82479000E-01
	(-0.9572E-01 J 0.3404E 00 ) E	0.8000E 01	0.13089931E 00
	(-0.7213E-01 J 0.2060E 00 ) T	0.9000E 01	0.18865520E 00
	(-0.9572E-01 J 0.3404E 00 ) E	0.1000E 02	0.23228356E 00
	(-0.1079E 01 J -0.3880E 00 ) T	0.1100E 02	0.31920141E 00
	(-0.5657E-01 J -0.4702E-01 ) E	0.1200E 02	0.38815959E 00
	(-0.1079E 01 J -0.3880E 00 ) T	0.1300E 02	0.44752681E 00
	(-0.5657E-01 J -0.4702E-01 ) E	0.1400E 02	0.50393325E 00
	(-0.1079E 01 J 0.3880E 00 ) T	0.1500E 02	0.55282429E 00
	(-0.5657E-01 J 0.4702E-01 ) E	0.1600E 02	0.59245241E 00
	(-0.1320E 01 J -0.9202E-12 ) T	0.1700E 02	0.62283635E 00
	(-0.5480E-01 J -0.6949E-08 ) E	0.1800E 02	0.64370114E 00
	(-0.1320E 01 J -0.9202E-12 ) T	0.1900E 02	0.65540731E 00
	(-0.5480E-01 J -0.6949E-08 ) E	0.2000E 02	0.65867203E 00
	(0.0000E 00 J 0.0000E 00 ) T	0.2100E 02	0.65448737E 00
	(0.0000E 00 J 0.0000E 00 ) T	0.2200E 02	0.64404070E 00
	(0.5000E 00 J 0.9109E-07 ) E	0.2300E 02	0.62863541E 00
		0.2400E 02	0.60981848E 00
		0.2500E 02	0.58831459E 00
		0.2600E 02	0.56597197E 00
		0.2700E 02	0.54371685E 00
		0.2800E 02	0.52252042E 00
		0.2900E 02	0.50317657E 00
		0.3000E 02	0.48362908E 00
		0.3100E 02	0.47227865E 00
		0.3200E 02	0.46137297E 00
		0.3300E 02	0.45363620E 00
		0.3400E 02	0.44898993E 00
		0.3500E 02	0.44721684E 00
		0.3600E 02	0.44801658E 00
		0.3700E 02	0.45100296E 00
		0.3800E 02	0.45575190E 00
		0.3900E 02	0.46191697E 00
		0.4000E 02	0.46815346E 00
		0.4100E 02	0.47613794E 00
		0.4200E 02	0.46328402E 00
		0.4300E 02	0.49075466E 00
		0.4400E 02	0.49737012E 00
		0.4500E 02	0.50321269E 00
		0.4600E 02	0.50812629E 00
		0.4700E 02	0.51202422E 00
		0.4800E 02	0.51486546E 00
		0.4900E 02	0.51666020E 00
		0.5000E 02	0.51749223E 00
			at t = 50 sec overshoot = 3.5%

	STEP	RESPONSE
	-3.40E-01 -2.40E-01 -1.40E-01 -4.00E-02 6.00E-02 1.60E-01 2.60E-01 3.60E-01 4.60E-01 5.60E-01 6.60E-01	
0.0000E 00	.....w.....	.....
	* * * *	*
	* * * *	*
	* * * *	*
5.0000E 00	.....*	.....
	* * * *	*
	* * * *	*
1.0000E 01	.....*	.....
	* * * *	*
	* * * *	*
1.5000E 01	.....*	.....
	* * * *	*
	* * * *	*
2.0000E 01	.....*	.....
	* * * *	*
	* * * *	*
2.5000E 01	.....*	.....
GET		
3.0000E 01	.....*	.....
	* * * *	*
	* * * *	*
3.5000E 01	.....*	.....
	* * * *	*
	* * * *	*
4.0000E 01	.....*	.....
	* * * *	*
	* * * *	*
4.5000E 01	.....*	.....
	* * * *	*
	* * * *	*
4.9999E 01	.....*	.....

NASAP RAMP RESPONSE	ERRCH	RSPONSE	RAMP	RESPONSE
NONE			TIME	IR3/VV1
RAMP RESPONSE			0.0000E 00	0.10728836E-05
V1 1 2 1			0.3000E-01	0.29993057E-01
R1 2 3 1			0.6000E-01	0.59932828E-01
C1 3 1 1F			0.9000E-01	0.89774251E-01
V2 1 4 1 VC1			0.1200E 00	0.11947465E 00
R2 4 5 0.5			0.1500E 00	0.14894968E 00
C2 5 1 1F			0.1800E 00	0.17830014E 00
V3 1 6 1 VC2			0.2100E 00	0.20735681E 00
R3 2 6 1	1= 4		0.2400E 00	0.23613775E 00
OUTPUT	2= 3		0.2700E 00	0.26461649E 00
IR3/VV1	3= 1		0.3000E 00	0.29277182E 00
TIME 1.5			0.3300E 00	0.32058012E 00
EXECUTE			0.3600E 00	0.34802485E 00
TRANSFER FUNCTION IR3/VV1			0.3900E 00	0.37509120E 00
(			0.4200E 00	0.40176356E 00
( 0.00E 00 +3.00E 00 S +1.00E 00 S ) <sup>2</sup>			0.4500E 00	0.42803073E 00
H(S) = 1.000E 00-----			0.4800E 00	0.45388103E 00
(			0.5100E 00	0.47930789E 00
( 2.00E 00 +3.00E 00 S +1.00E 00 S ) <sup>2</sup>			0.5400E 00	0.50430220E 00
ZERO_OF TRANSFER FUNCTION			0.5700E 00	0.52885991E 00
ZERO REAL PART IMAG. PART			0.6000E 00	0.55297475E 00
POLE OF TRANSFER FUNCTION			0.6300E 00	0.57664478E 00
POLE REAL PART IMAG. PART			0.6600E 00	0.59986609E 00
1 0.0000E 00 0.0000E 00	1 -0.1000E 01 0.0000E 00		0.6900E 00	0.62263088E 00
2 -0.3000E 01 0.0000E 00	2 -0.2000E 01 0.0000E 00		0.7200E 00	0.64495999E 00
RAMP RESPONSE FUNCTION			0.7500E 00	0.66683275E 00
F(T) =			0.7800E 00	0.68825656E 00
(-0.2000E 01 J-0.4210E-09 ) E			0.8100E 00	0.70923400E 00
(-0.1000E 01 J 0.0000E 00 ) T			0.8400E 00	0.72976661E 00
( 0.5000E 00 J 0.5262E-10 ) E			0.8700E 00	0.74987778E 00
( 0.0000E 00 J 0.1000E-02 ) T			0.9000E 00	0.76951081E 00
( 0.7500E 00 J-0.8750E-03 ) E			0.9300E 00	0.78972955E 00
( 0.0000E 00 J-0.1000E-02 ) T			0.9600E 00	0.80751842E 00
( 0.7500E 00 J 0.8750E-03 ) E			0.9900E 00	0.82588164E 00
			0.1020E 01	0.84382498E 00
			0.1050E 01	0.86135292E 00
			0.1080E 01	0.87847179E 00
			0.1110E 01	0.89518672E 00
			0.1140E 01	0.91150403E 00
			0.1170E 01	0.92742980E 00
			0.1200E 01	0.94297022E 00
			0.1230E 01	0.95813265E 00
			0.1260E 01	0.97292124E 00
			0.1290E 01	0.98734286E 00
			0.1320E 01	0.10014066E 01
			0.1350E 01	0.10151215E 01
			0.1380E 01	0.10284872E 01
			0.1410E 01	0.10419154E 01
			0.1440E 01	0.10542107E 01
			0.1470E 01	0.10665580E 01
			0.1500E 01	0.10786314E 01
			(-0.2000E 01 J 0.0000E 00 ) T	
			( 0.5000E 00 J 0.5262E-10 ) E	
			( 0.0000E 00 J 0.1000E-02 ) T	
			( 0.7500E 00 J-0.8750E-03 ) E	
			( 0.0000E 00 J-0.1000E-02 ) T	
			( 0.7500E 00 J 0.8750E-03 ) E	
				20v & = 0.001
				A = 0.75
				A >> B
				B = 0.875 x 10 <sup>-3</sup>
				20v Bt + 2A = 1.5

20v 5.74

## RAMP

## RESPONSE

	-9.20E-01	-7.20E-01	-5.20E-01	-3.20E-01	-1.20E-01	8.00E-02	2.80E-01	4.80E-01	6.80E-01	8.80E-01	1.08E 00
--	-----------	-----------	-----------	-----------	-----------	----------	----------	----------	----------	----------	----------

0.0000E 00	.....	.....	.....	.....	.....	*	.....	.....	.....	.....	.....
	.	.	.	.	.	*	I	.	.	.	.
	.	.	.	.	.	*	I	.	.	.	.
	.	.	.	.	.	*	I	.	.	.	.
1.5000E-01	.....	.....	.....	.....	.....	*	I	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
3.0000E-01	.....	.....	.....	.....	.....	*	I	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
4.5000E-01	.....	.....	.....	.....	.....	*	I	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
6.0000E-01	.....	.....	.....	.....	.....	*	I	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
7.5000E-01	.....	.....	.....	.....	.....	*	I	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
9.0000E-01	.....	.....	.....	.....	.....	*	I	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
1.0500E 00	.....	.....	.....	.....	.....	*	I	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
1.2000E 00	.....	.....	.....	.....	.....	*	I	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
1.3500E 00	.....	.....	.....	.....	.....	*	I	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
	.	.	.	.	.	I	.	*	.	.	.
1.5000E 00	.....	.....	.....	.....	.....	*	I	*	.	.	.

212

245.7°

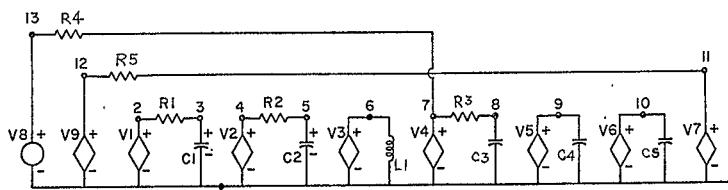
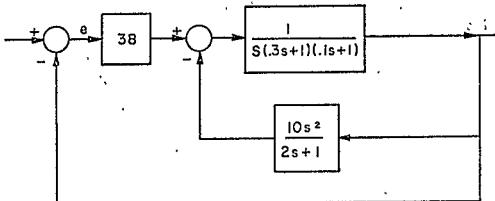
We continue with the Newton Gould and Kaiser problem, see Fig. 4.9, to illustrate error responses. In Fig. 5.8 gives the step error. Figure 5.9 gives the step response. The latter is included here to emphasize that only a single NASA computer card need be changed to get the alternative response output. Finally in Fig. 5.10 we show the ramp error response for this same control system.

RADIAN  
STEP

```

V8 1 13 1.
R4-13 -7 1.
V9 1 12 38 IR4
R5-12 11 1.
V1 1 2 1. IR5
R1-2 3 0 3
C1 3 1 1P
V2-1 1 1. -VG1
R2 4 5 0 1
C2 5 1 1P
V3 1 6 1. VC2
L1-6 1 1H
V4 1 7 1. IL1
R3 7 8 2
C3 8 1 1P
V5 1 9 1. VC3
C4 9 1 1P
V6-1 10 10 IC4
C5 10 1 1F
V7-11 1 1G5
OUTPUT
IR4/VV8/V9
FREQ-1.0 2.0 0.05
TIME-2.0
EXECUTE

```



TRANSFER FUNCTION IR4/VV8/V9

$$\left( \begin{array}{cccccc} 0.00E+00 & +1.67E-01 s & -1.207E-02 s^2 & +1.30E-01 s^3 & +1.00E-00 s^4 \end{array} \right)$$

 $H(s) = 1.000E-00$ 

$$\left( \begin{array}{cccccc} 6.33E-02 & +1.28E-03 s & +2.07E-02 s^2 & +1.30E-01 s^3 & +1.00E-00 s^4 \end{array} \right)$$

## ZERO-OF-TRANSFER-FUNCTION

ZERO REAL PART IMAG. PART

1	0.00000E 00	0.00000E 00
2	-0.81089E-01	-0.00090E-00
3	-0.68761E 01	0.12581E 02
4	-0.60761E-03	0.12501E 02

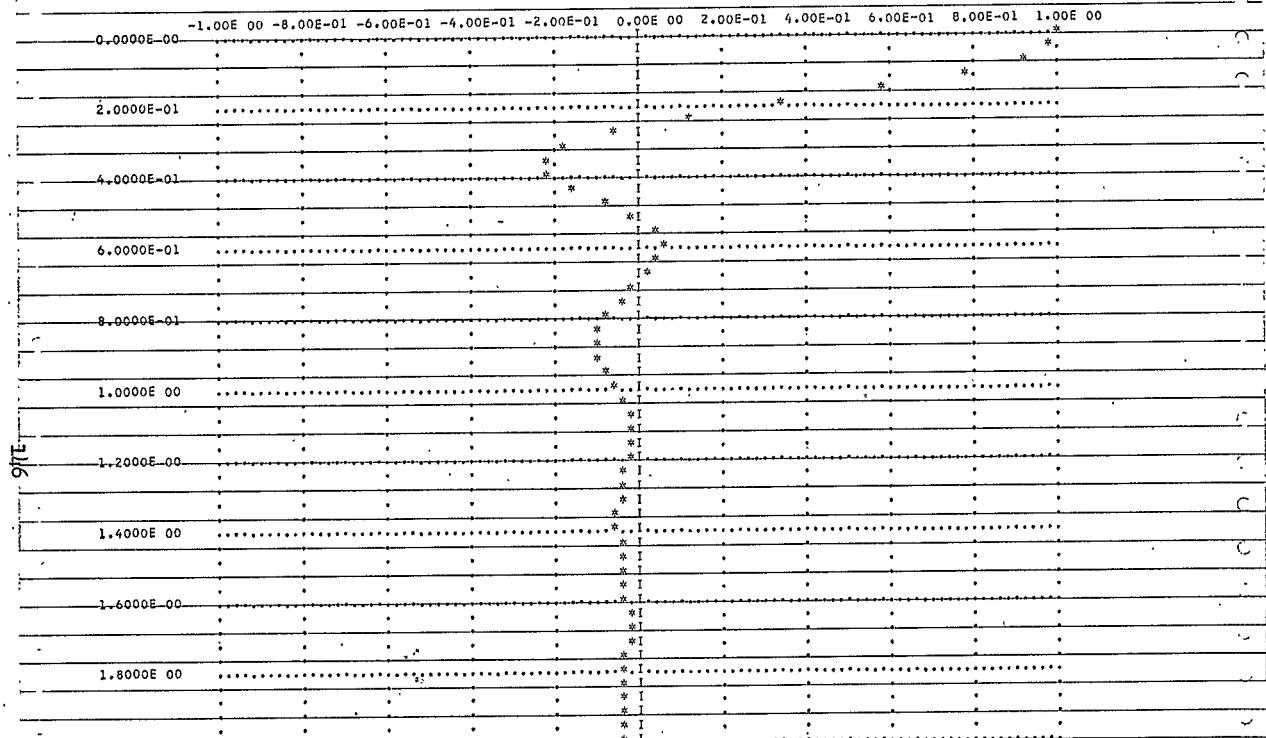
## POLE OF TRANSFER FUNCTION

POLE	REAL PART	IMAG. PART
1	-0.53860E-00	0.00000E-00
2	-0.28810E 01	0.12157E 02
3	-0.20815E-01	0.12157E-02
4	-0.75327E 01	0.00000E 00

STEP	RESPONSE FUNCTION	STEP	RESPONSE
F(T) =		TIME	IR4/VVB
	( -0.8468E-01 ) E	0.0000E-00	-0.89999570E-00
	( -0.1472E-07 ) E	0.4000E-01	0.98830140E-00
	( -0.5386E 00 ) J 0.0000E 00 ? T	0.8000E-01	0.93671491E-00
	( -0.2881E-01 ) J-0.1216E-02 ) T	0.1200E-00	0.77826107E-00
	( 0.4340E-01 J-0.3176E 00 ) E	0.1600E-00	0.87408259E-00
	( -0.2881E 01 J-0.1216E 02 ? T	0.2000E 00	0.34088778E-00
	( -0.4340E-01 J 0.3176E-00 ) E	0.2400E-00	0.17464548E-00
	( -0.7533E-01 J-0.0000E-00 ) T	0.2800E-00	-0.62708139E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.3200E-00	-0.17854778E-00
	( 0.0000E 00 J 0.0000E 00 ? T	0.3600E-00	-0.22616476E-00
	( 0.6000E-00 J 0.0000E-00 ) E	0.4000E-00	-0.23346736E-00
	( -0.7533E-01 J-0.0000E-00 ) T	0.4400E-00	-0.15973060E-00
	( 0.9979E 00 J-0.2777E-07 ) E	0.4800E-00	-0.87481629E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.5200E-00	-0.19262429E-01
	( 0.6000E-00 J 0.0000E-00 ) E	0.5600E-00	-0.20723268E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.6000E-00	0.53309571E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.6400E-00	-0.49267274E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.6800E-00	0.24339374E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.7200E-00	-0.11551964E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.7600E-00	-0.49438279E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.8000E-00	-0.76292327E-01
	( 0.6000E-00 J 0.0000E-00 ) E	0.8400E-00	-0.97540200E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.8800E-00	-0.10153145E-00
	( 0.9979E 00 J-0.2777E-07 ) E	0.9200E-00	-0.93511110E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.9600E-00	-0.77546101E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.1000E-01	-0.58880298E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.1040E-01	-0.41192196E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.1080E-01	-0.28661814E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.1120E-01	-0.22530064E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.1160E-01	-0.22621151E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.1200E-01	-0.27435135E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.1240E-01	-0.34752585E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.1280E-01	-0.22242238E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.1320E-01	-0.48091702E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.1360E-01	-0.51122818E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.1400E-01	-0.31082220E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.1440E-01	-0.48424244E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.1480E-01	-0.44095926E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.1520E-01	-0.39202157E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.1560E-01	-0.34757502E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.1600E-01	-0.31470537E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.1640E-01	-0.29654402E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.1680E-01	-0.29240642E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.1720E-01	-0.29869609E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.1760E-01	-0.31024324E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.1800E-01	-0.32216437E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.1840E-01	-0.33604209E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.1880E-01	-0.33160023E-01
	( 0.0000E 00 J 0.0000E 00 ? T	0.1920E-01	-0.32636251E-01
	( -0.7533E-01 J-0.0000E-00 ) T	0.1960E-01	-0.31546666E-01
	( 0.9979E 00 J-0.2777E-07 ) E	0.2000E-01	-0.30108280E-01

## STEP

## RESPONSE



STEP	RESPONSE FUNCTION	STEP	RESPONSE
F(T) =		TIME	VV4/VVB
( -0.8468E-01 J 0.0000E 00 ) E	( -0.5386E 00 J 0.0000E 00 ) T	0.0000E_00	0.0000000E_00
( -0.4340E-01 J 0.3176E 00 ) E	( -0.2881E-01 J -0.1216E-02 ) T	0.4000E_01	0.11698127E-01
( -0.4340E-01 J 0.3176E 00 ) E	( -0.2881E 01 J -0.1216E 02 ) T	0.8000E_01	0.79284728E-01
( -0.9979E 00 J 0.3546E-07 ) E	( -0.7533E-01 J -0.0000E 00 ) T	0.1200E_00	0.22173876E_00
( -0.1000E-01 J -0.6523E-07 ) E	( 0.0000E 00 J 0.0000E 00 ) T	0.1600E_00	0.42591140E_00
		0.2000E_00	0.65911257E_00
		0.2400E_00	0.88253599E_00
		0.2800E_00	0.10627079E_01
		0.3200E_00	0.11789455E_01
		0.3600E_00	0.12761648E_01
		0.4000E_00	0.12123467E_01
		0.4400E_00	0.11597385E_01
		0.4800E_00	0.10878811E_01
		0.5200E_00	0.10192623E_01
		0.5600E_00	0.96566694E_00
		0.6000E_00	0.94669074E_00
		0.6400E_00	0.95073295E_00
		0.6800E_00	0.97566698E_00
		0.7200E_00	0.10119514E_01
		0.7600E_00	0.10494385E_01
		0.8000E_00	0.10736938E_01
		0.8400E_00	0.10975399E_01
		0.8800E_00	0.11115310E_01
		0.9200E_00	0.10935106E_01
		0.9600E_00	0.10775442E_01
		0.1000E_01	0.10585804E_01
		0.1040E_01	0.10411921E_01
		0.1080E_01	0.10286617E_01
		0.1120E_01	0.10225296E_01
		0.1160E_01	0.10226212E_01
		0.1200E_01	0.10274353E_01
		0.1240E_01	0.10347528E_01
		0.1280E_01	0.10422621E_01
		0.1320E_01	0.10480919E_01
		0.1360E_01	0.10511227E_01
		0.1400E_01	0.10510817E_01
		0.1440E_01	0.10684276E_01
		0.1480E_01	0.10440960E_01
		0.1520E_01	0.10392017E_01
		0.1560E_01	0.10347576E_01
		0.1600E_01	0.10314703E_01
		0.1640E_01	0.10296545E_01
		0.1680E_01	0.10292397E_01
		0.1720E_01	0.10298691E_01
		0.1760E_01	0.10310245E_01
		0.1800E_01	0.10322161E_01
		0.1840E_01	0.10330038E_01
		0.1880E_01	0.10331602E_01
		0.1920E_01	0.10326357E_01
		0.1960E_01	0.10315466E_01
		0.2000E_01	0.10301085E_01

LTT

**STEP RESPONSE**

A scatter plot showing the relationship between TITI (Y-axis) and TITE (X-axis). The X-axis (TITE) ranges from -0.780 to 1.220. The Y-axis (TITI) ranges from 0.0000E+00 to 2.0000E+00. The data points, represented by asterisks (\*), show a strong positive linear correlation, starting near (-0.780, 0.0000) and ending near (1.220, 2.0000).

TITE	TITI
-0.7800	0.0000
-0.5800	0.0000
-0.3800	0.0000
-0.1800	0.0000
0.0000	0.0000
0.2000	0.0000
0.4000	0.0000
0.6000	0.0000
0.8000	0.0000
1.0000	0.0000
1.2000	0.0000
-0.7800	0.2000
-0.5800	0.2000
-0.3800	0.2000
-0.1800	0.2000
0.0000	0.2000
0.2000	0.2000
0.4000	0.2000
0.6000	0.2000
0.8000	0.2000
1.0000	0.2000
1.2000	0.2000

NASA-P CLARK PEGG  
NEWTON-G-K

RADIANS  
RAMP-RESPONSE

V8 1 13 1,  
R4 1 9 -1,

V9 1 12 38 IR4

R5 2 11 1,

V1 1 2 1, IR5

R3 2 3 0 3

C1 3 1 1F

V2 1 4 1, V61

R2 4 5 0 1

G2 5 1 1F

V3 1 6 1, VC2

L1 6 1 1H

V4 1 7 1, IL1

R3 7 8 2,

C3 8 1 1F

V5 1 9 1, V63

C4 9 1 1F

V6 1 10 20 -IG4

C5 10 1 1F

V7 1 14 1, IG5

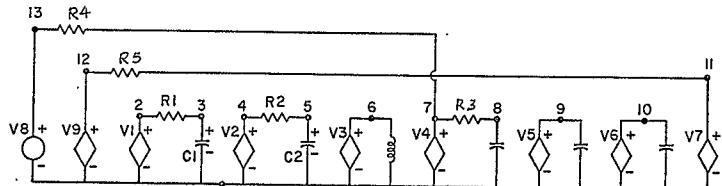
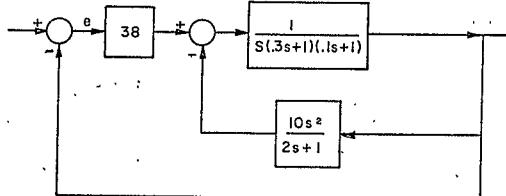
OUTPUT

IR4/VV8/V9

TIME 2.0

EXECUTE

# ERROR RESPONSE



TRANSFER FUNCTION IR4/VV8/V9

$$H(s) = \frac{1 + 0.00E + 00}{(6.33E - 02 + 1.28E - 03 s + 2.07E - 02 s^2 + 1.38E - 01 s^3 + 1.00E - 00 s^4)}$$

NUMBER OF LOOPS PER ORDER

1= 6

2= 8

3= 4

4= 1

ZERO-OF-TRANSFER-FUNCTION

ZERO REAL PART IMAG. PART

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

1 0.00000E 00 0.00000E 00

2 -0.81083E -01 0.00000E -00

3 -0.68761E 01 0.12581E 02

4 -0.68761E 01 -0.12581E 02

1 -0.53060E -00 0.00000E -00

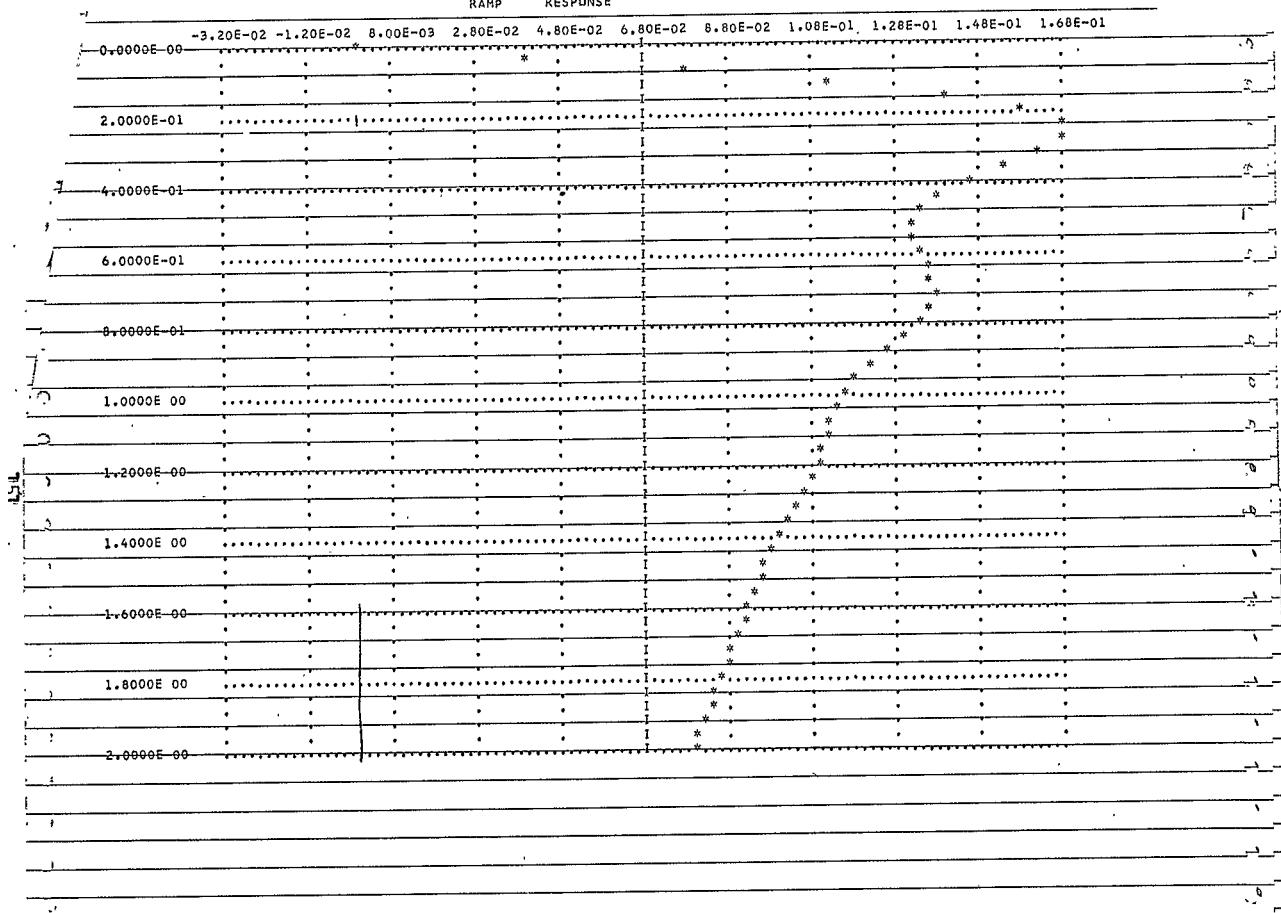
2 -0.28810E 01 0.12157E 02

3 -0.20810E 01 0.12157E 02

4 -0.75327E 01 0.00000E 00

RAMP	RESPONSE FUNCTION	RAMP	RESPONSE
F(T) =		TIME	IR4/VV8
	( -0.5386E 00 J 0.0000E 00 ) T	0.0000E-00	0.31956387E-07
	( -0.1572E 00 J -0.2739E-07 ) E	0.4000E-01	0.39879285E-01
	( -0.2553E-01 J 0.2481E-02 ) E	0.8000E-01	0.74200936E-01
	( -0.2881E-01 J -0.1216E-02 ) T	0.1200E 00	0.11252379E 00
	( -0.2881E 01 J -0.1216E 02 ) T	0.1600E-00	0.13972344E-00
	( -0.2553E-01 J -0.2481E-02 ) E	0.2000E 00	0.15806085E 00
	( -0.7533E-01 J -0.0000E-00 ) T	0.2400E-00	0.16713339E-00
	( -0.1325E 00 J 0.3688E-08 ) E	0.2800E 00	0.16894079E 00
	( 0.0000E 00 J 0.1000E-02 ) T	0.3200E-00	0.16297650E 00
	( -0.1316E-01 J 0.1365E-03 ) E	0.3600E 00	0.15465176E 00
	( -0.0000E-00 J -0.1000E-02 ) T	0.4000E-00	0.14561663E-00
	( 0.1316E-01 J -0.1365E-03 ) E	0.4400E 00	0.13812327E 00
		0.4800E-00	0.13314849E-00
		0.5200E 00	0.13104695E 00
		0.5600E-00	0.13125147E-00
		0.6000E 00	0.13312036E 00
		0.6400E-00	0.13525462E-00
		0.6800E 00	0.13678169E 00
		0.7200E-00	0.13705003E-00
		0.7600E 00	0.13781055E 00
		0.8000E-00	0.13219290E-00
		0.8400E 00	0.12960213E 00
		0.8800E-00	0.12557584E-00
		0.9200E 00	0.12164086E 00
		0.9600E-00	0.11820338E-00
		0.1000E 01	0.11547697E 00
		0.1044E 01	0.11349236E-00
		0.1080E 01	0.11211604E 00
		0.1120E-01	0.11111414E-00
		0.1160E 01	0.11023003E 00
		0.1200E-01	0.10924119E-00
		0.1240E 01	0.10800177E 00
		0.1280E 01	0.10645354E 00
		0.1320E 01	0.10464364E 00
		0.1360E-01	0.10264891E-00
		0.1400E 01	0.10059488E 00
		0.1440E-01	0.98597368E-01
		0.1480E 01	0.96743226E-01
		0.1520E-01	0.95077157E-01
		0.1560E 01	0.93600669E-01
		0.1600E-01	0.92280924E-01
		0.1640E 01	0.91063499E-01
		0.1680E 01	0.89889765E-01
		0.1720E 01	0.88710368E-01
		0.1760E 01	0.87493122E-01
		0.1800E 01	0.86227357E-01
		0.1840E 01	0.84921122E-01
		0.1880E 01	0.83595574E-01
		0.1920E 01	0.82277477E-01
		0.1960E 01	0.80992222E-01
		0.2000E 01	0.79758406E-01

## RAMP      RESPONSE



#### VD. FIGURES OF MERIT BASED ON ERROR SIGNAL

The conventional criteria on this basis are the transient performance and the steady state performance. These were briefly summarized in Chapter IV.

The overall response of a control system is determined by the poles and zeros of its transfer function  $G_f(s) = N_f(s) / D_f(s)$ ; it is not possible to tell it from the degree of  $D_f(s)$  and the difference of the degrees of  $D_f(s)$  and  $N_f(s)$  alone. However it is found that for the same  $\delta N_f$  an optimal system with a smaller  $\delta D_f$  is better than such a system with a larger  $\delta D_f$ ; where we have let  $\delta(\cdot)$  denote the degree of the polynomials. Similarly for the same  $\delta D_f$ , an optimal system with a smaller ( $\delta D_f - \delta N_f$ ) is better than such a system with a larger ( $\delta D_f - \delta N_f$ ). Therefore, given a plant, for which we can choose the degree of  $\delta D_f$  and  $\delta N_f$ , it is desirable to choose a smaller  $\delta D_f$  and a smaller ( $\delta D_f - \delta N_f$ ). Now the smallest possible ( $\delta D_f - \delta N_f$ ) is governed by the given plant.

It can be shown [CH 1] that the absolute minimum of  $\delta D_f$  is  $\delta D - \delta N$ ; where D and N refer to the uncompensated system. To achieve this minimum,  $\delta N_f$  is required to be zero. In general  $\delta N_f \geq \delta N$ , unless some pole-zero cancellations are employed in the design.

The transient performance of a system may be specified by percentage overshoot, rise time and settling time. These specifications are dictated by the poles and zeros of the transfer function. Since we do not have control over the zeros of a system, usually we just try to put the poles of the over-all system in some desired location. For a second order transfer function with a constant numerator, the desired pole locations can be readily determined from the transient specifications. For high order transfer functions, the concept of dominant poles can often be used. Then a pair of complex conjugate poles is located as in the second order transfer function and the rest of the poles are located in the far left half plane with real parts at least ten times as large as the real parts of the conjugate poles. In choosing these poles, the steady state performance should be kept in mind.

The steady-state performance of a control system  $G_f(s)$  depends only on the coefficients of  $G_f(s)$ . Let

$$G_f(s) = \frac{b_0 + b_1 s + b_2 s^2 + \dots + b_{n-1} s^{n-1}}{a_0 + a_1 s + a_2 s^2 + \dots + a_{n-1} s^{n-1} + s^n},$$

If the steady-state error due to a step input and a ramp input are required to be smaller than  $k^{\alpha}/b$ , then we need, respectively,

$$\left| \frac{b_0 - a_0}{a_0} \right| \leq k/100; \quad \text{and} \quad a_0 = b_0, \quad \left| \frac{b_1 - a_1}{a_0} \right| \leq k/100.$$

Hence the steady-state performance of a system can be rather easily controlled. When using these criteria it is necessary to check the response of the chosen overall transfer function with an analog or a digital computer to be sure that it is satisfactory before continuing the design.

In addition to the conventional criteria just discussed, we shall mention a few other criteria based on the error signal that serve to make the control system "optimum" in some sense.

(i) ITAE criterion (Integral of time-multiplied absolute-value of error):

This criterion was first introduced by Graham and Lathrop [GA 1]. For a given plant, the problem is to design an over-all system which minimizes

$$\int_0^\infty t|r(t) - c(t)|dt$$

where  $r$  is the reference or desired signal and  $c$  is the output of the over-all system. It is clear that  $|r(t) - c(t)|$  is the error between the desired signal and the actual output. The multiplication of  $t$  on  $|r(t) - c(t)|$  provides an increasingly heavy penalty for a sustained error. Using step functions as reference inputs, Graham and Lathrop obtained, by analog computer simulation a set of optimal transfer functions [GA 1].

To choose an optimal transfer function  $G_p(s)$  for a given plant  $G(s)$  note that if all the forward paths from  $r$  to  $c$  pass through the plant, then the zeros of the plant (the roots of  $N(s)$ ) will be independent of how the compensators are introduced. Consequently if the numerator of  $G_p(s)$  does not contain all the zeros of  $G(s)$ , the missing zero must be cancelled by a pole. The advantage of this criterion is that it is very selective; however it cannot be studied analytically. This criterion is not widely used, because a complete list of optimal transfer functions is not available.

- (ii) Quadratic criterion: The optimal system is the one which minimizes

$$\int_0^\infty [u^2(t) + (r(t) - c(t))^2] dt$$

where  $u$  is the input to the plant. In this criterion, if  $u^2$  is not included, the optimal transfer function will always be unity and the required compensators may not be physically realizable; furthermore the magnitude of  $u$  may be large and the system will be saturated. The optimal transfer function for a given plant and a given reference input  $r$  can be obtained by applying Chang's root-square-locus method [CH 1] as well as by using the dynamical equation description [see AT 1]

The design by using the quadratic criterion can be solved rigorously. However there are three arguments against using this criterion. First, it is not very selective [GR 1]. Second, the criterion is chosen mainly for mathematical convenience rather than practical reasons. Finally and most seriously, the resulting optimal transfer function may not be realizable in practice. If all the zeros of the transfer function of the plant  $G(s) = \frac{N(s)}{D(s)}$  have negative real parts and if the reference input is a step function, the optimal

transfer function is of the form  $\frac{K(s)}{D_f(s)}$  and  $\delta D_f(s) = \delta D$ . In this case, it is easily realized. However for plants with nonminimum-phase transfer function, the design by using quadratic criterion might have difficulty. Another difficulty in using this criterion occurs when the reference input is a ramp.

- (iii) ISE criterion (the integral of the error squared): Here one seeks to minimize

$$\int_0^t e^2(t) dt.$$

To obtain any of these criteria from NASAP use of an external integration subroutine is required. To illustrate this we follow Beck [BE 1] and use the integral of the squared error as the performance index although other criteria can be applied with equal ease since an analytical solution is not required. The formation of the chosen performance index is indicated in Fig. 5.7 taken from [BE 1] where it is shown that the

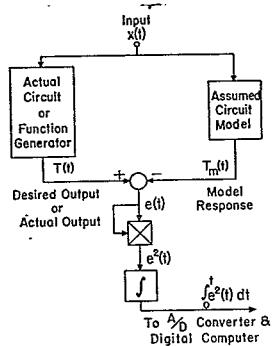


Fig. 5.7: Index of performance formation

ISE is generated in the analog computer and returned to the digital computer through the analog to digital converters. In the hybrid computer application an optimization algorithm operates upon this output.

## CHAPTER VI

## SENSITIVITY ANALYSIS

## VIA INTRODUCTION TO SENSITIVITY

The NASAP sensitivity results, given in tabular and graphical form, can be used to predict the percent change, absolute change, and modified value of the transfer function to changes in a particular network parameter. By definition the sensitivity of some real function  $\alpha$  (in NASAP,  $R_{EH}$ ,  $|H|$  and  $\phi$ ) to a change in a real parameter  $x$  (in NASAP, resistance, capacitance, inductance and dependency value) is defined as

$$S_x^\alpha \triangleq \frac{x}{\alpha} \frac{d\alpha}{dx} = \frac{\frac{d\alpha}{dx}}{\frac{\alpha}{x}} = \frac{d\ln\alpha}{d\ln x} \quad (6.1)$$

By rearranging (6.1), the differential  $d\alpha$  can be related to the differential  $dx$ ,

$$d\alpha = \alpha \sum_x^\alpha \frac{dx}{x} \quad (6.2)$$

For an incremental change,  $\Delta x$ , in the parameter  $x$  the incremental change,  $\Delta \alpha$ , in  $\alpha$  can be approximated from (6.2)

$$\Delta \alpha \approx \alpha \sum_x^\alpha \frac{\Delta x}{x} \quad (6.3a)$$

If the change in  $x$  is expressed as some percentage of  $x$ , then (6.3a) can be expressed as

$$\Delta \alpha \approx \alpha \sum_x^\alpha \frac{Px}{100} \quad (6.3b)$$

where  $Px$  is the percent change in the parameter  $x$ . The percent change in  $\alpha$  is easily found by dividing both sides of (6.3b) by  $\alpha$ , thus

$$P\alpha \approx \sum_x^\alpha Px \quad (6.4)$$

where  $P_{\alpha}$  is the percent change in the function  $\alpha$ . The modified value of  $\alpha$ , called  $\alpha'$ , can be expressed as

$$\alpha' \stackrel{\Delta}{=} \alpha + \frac{P_{\alpha}}{100} \alpha \quad (6.5)$$

$$\alpha' \approx \alpha \left( 1 + \sum_x \frac{\Delta x}{x} \right) \quad (6.6a)$$

or

$$\alpha' \approx \alpha \left( 1 + \sum_x \frac{P_x}{100} \right) \quad (6.6b)$$

Naturally the smaller  $\Delta x$  or  $P_x$ , the more accurate  $\Delta \alpha$ ,  $\alpha'$ , and  $P_{\alpha}$  will be. The accuracy will also be enhanced if  $\alpha$  is roughly a linear function of  $x$  over the range of  $x$  under investigation.

If indeed  $\alpha$  is a linear function of  $x$  of the form

$$\alpha = Kx, \quad (6.7)$$

then from (6.1) it is seen that

$$\sum_x \alpha = 1 \quad (6.8)$$

Furthermore from (6.1) if  $\alpha$  is independent of  $x$ , then

$$\sum_x \alpha = 0 \quad (6.9)$$

or the function  $\alpha$  is insensitive to changes in  $x$ .

Similarly the root sensitivity printed out by NASA/P can be used to predict the new pole and zero locations for a small change in some network parameter. The root sensitivity is defined as

$$\sigma_x^p \triangleq x \frac{dp}{dx} \quad (6.10)$$

where  $p$  is a zero or pole of the given transfer function. The differential  $dp$  can be expressed in terms of the root sensitivity as

$$dp = \sigma_x^p \frac{dx}{x} = \sigma_x^p d \ln x \quad (6.11)$$

Note that  $dp$  will be complex since  $\sigma_x^p$  is also a complex number while  $\frac{dx}{x}$  is a real quantity.

The incremental change in  $p$ ,  $\Delta p$ , for an incremental change in  $x$  is derived from (6.11)

$$\Delta p \approx \sigma_x^p \frac{\Delta x}{x} \quad (6.12)$$

If  $\Delta x$  is expressed as some percentage of  $x$  then (6.12) becomes

$$\Delta p \approx \sigma_x^p \frac{Px}{100} \quad (6.13)$$

where  $Px$  is the percent change in  $x$ .

#### VIB DERIVATION OF SENSITIVITY FORMULAS

Although the formulas used in NASAP to calculate the transfer function sensitivity were derived from a tagging technique on the loops of the flow-graph (see [MA 1]), these sensitivity formulas can be obtained by using simple calculus. Suppose the transfer function is given

$$H(s) = \frac{N(s)}{D(s)} \quad (6.14)$$

where  $N(s)$  and  $D(s)$  are polynomials in  $s$ .

Since  $H(s)$  is the transfer function of a linear circuit it can be expressed as a bilinear function of any element in the circuit. That is, both

$N(s)$  and  $D(s)$  can be expressed as the sum of two polynomials in  $s$  where one polynomial does and the other does not contain the specified element.

Thus

$$H(s) = \frac{N(s)}{D_1(s)} = \frac{A(s) + x B(s)}{C(s) + x D(s)} \quad (6.15)$$

where  $x$  is the specified element in the circuit and  $A, B, C$ , and  $D$  are polynomials in  $s$ .

The sensitivity of the transfer function  $H(s)$  to some parameter  $x$  is defined as

$$\sum_x H(s) \xrightarrow{\Delta \frac{dH}{dx}/x} = \frac{d \ln H}{d \ln x} \quad (6.16)$$

Thus to find the sensitivity of  $H(s)$  to changes in  $x$  in (6.15), one obtains

$$\sum_x H(s) = \sum_x N(s) - \sum_x D_1(s) \quad (6.17)$$

but

$$\begin{aligned} \sum_x N(s) &= \frac{x}{N(s)} \frac{d N(s)}{dx} \\ &= \frac{x}{A+Bx} \frac{d}{dx} (A+Bx) \\ &= \frac{Bx}{A+Bx} \end{aligned} \quad (6.18)$$

Similarly

$$\sum_x D_1(s) = \frac{Dx}{C+Dx} \quad (6.19)$$

Substituting (6.18) and (6.19) into (6.17) yields

$$\sum_x H(s) = \frac{Bx}{A+Bx} - \frac{Dx}{C+Dx} \quad (6.20)$$

This expression remains unchanged if 1 is added and subtracted from the righthand side. Then we can write

$$\begin{aligned}
 S_x^{H(s)} &= \frac{Bx}{A+Bx} - \frac{A+Bx}{A+Bx} - \frac{Dx}{C+Dx} + \frac{C+Dx}{C+Dx} \\
 &= -\frac{A}{A+Bx} + \frac{C}{C+Dx} \\
 &= -\frac{A(s)}{N(s)} + \frac{C(s)}{D_1(s)}
 \end{aligned} \tag{6.21}$$

This equation can also be derived from the tagging technique on the loops of the flowgraph and is the one used in subroutine SENSS of NASAP to determine the transfer function sensitivity.

#### VIC DISCUSSION OF SENSITIVITY FORMULAS IN SENS

In subroutine SENS of the NASAP program are calculated the sensitivity expressions  $S_x^{\text{Re}H}$ ,  $S_x^{\text{Im}H}$ ,  $S_x^{|H|}$  and  $S_x^\phi$

where  $H(j\omega) = \text{Re}H(j\omega) + j\text{Im}H(j\omega)$

and  $H(j\omega) = |H(j\omega)|e^{j\phi}$ .

Again the tagging techniques of [1] are used to determine the sensitivity expressions. The basis for the tagging procedure is that the transfer function  $H(s)$  can be written as in (6.15) where  $x$  is the sensitivity parameter. Since the polynomials  $A$ ,  $B$ ,  $C$ , and  $D$  are, in general, complex quantities for  $s = j\omega$ , (6.15) can be rewritten as

$$H(j\omega) = \frac{(\text{Re} A + x \text{Re} B) + j(\text{Im} A + x \text{Im} B)}{(\text{Re} C + x \text{Re} D) + j(\text{Im} C + x \text{Im} D)} \tag{6.22}$$

where  $A(j\omega) = \text{Re}A(j\omega) + j\text{Im}A(j\omega)$  etc.

After some mathematical manipulation, the right side of (6.22) can be separated into its real and imaginary parts. Thus

$$\text{ReH}(j\omega) = \frac{N_R}{D} = \frac{(ReA \cdot ReC + ImA \cdot ImC) + x(ReB \cdot ReC + ReA \cdot ReD + ImB \cdot ImC + ImA \cdot ImD) + x^2(ReB \cdot ReD + ImB \cdot ImD)}{\left[(ReC)^2 + (ImC)^2\right] + 2x(ReC \cdot ReD + ImC \cdot ImD) + x^2\left[(ReD)^2 + (ImD)^2\right]} \quad (6.23)$$

and

$$\text{ImH}(j\omega) = \frac{N_I}{D} = \frac{(ImA \cdot ReC - ReA \cdot ImC) + x(ImA \cdot ReD - ReA \cdot ImD + ImB \cdot ReC - ReB \cdot ImC) + x^2(ImB \cdot ReD - ReB \cdot ImD)}{\left[(ReC)^2 + (ImC)^2\right] + 2x(ReC \cdot ReD + ImC \cdot ImD) + x^2\left[(ReD)^2 + (ImD)^2\right]} \quad (6.24)$$

Thus by use of (6.17) one obtains

$$\begin{aligned} S_x^{\text{ReH}} &= S_x^{N_R} - S_x^D \\ &= \frac{x}{N_R} \frac{d}{dx} N_R - \frac{x}{D} \frac{d}{dx} D \end{aligned}$$

or

$$S_x^{\text{ReH}} = \frac{x}{N_R D} \left( D \frac{d}{dx} N_R - N_R \frac{d}{dx} D \right) \quad (6.25)$$

$$\text{where } \frac{dN_R}{dx} = (ReB \cdot ReC + ReA \cdot ReD + ImB \cdot ImC + ImA \cdot ImD) + 2x(ReB \cdot ReD + ImB \cdot ImD) \quad (6.26)$$

$$\text{and } \frac{dD}{dx} = 2(ReC \cdot ReD + ImC \cdot ImD) + 2x \left[ (ReD)^2 + (ImD)^2 \right] \quad (6.27)$$

and  $N_R$  and  $D$  are defined in (6.23).

By a similar use of (6.17) one arrives at the expression

$$S_x^{\text{ImH}} = \frac{x}{N_I D} D \left( \frac{dN_I}{dx} - N_I \frac{dD}{dx} \right) \quad (6.28)$$

where  $\frac{dN_I}{dx} = (\text{ImA ReD} - \text{ReA ImD} + \text{ImB ReC} - \text{ReB ImC}) + 2x(\text{ImB ReD} - \text{ReB ImD}) \quad (6.29)$

and  $\frac{dD}{dx}$  is given, (6.27), and  $N_I$  and  $D$  are defined in (6.24).

Equations (6.25) and (6.28) are used in subroutine SENS to evaluate the sensitivities of the real part and imaginary part of the transfer function to changes in the parameter  $x$ .

The sensitivities  $S_x^{|H|}$  and  $S_x^\phi$  are evaluated in terms of the sensitivities found in (6.25) and (6.28). By definition

$$|H| = \sqrt{(\text{Re}H)^2 + (\text{Im}H)^2} \quad (6.30)$$

Differentiating this with respect to the sensitivity parameter  $x$  yields

$$\frac{d}{dx}|H| = \frac{\text{Re}H \frac{d\text{Re}H}{dx} + \text{Im}H \frac{d\text{Im}H}{dx}}{\sqrt{(\text{Re}H)^2 + (\text{Im}H)^2}} \quad (6.31)$$

By definition

$$S_x^{|H|} = \frac{x}{|H|} \frac{d|H|}{dx} \quad (6.32)$$

Substituting (6.30) and (6.31) into (6.32) results in

$$S_x^{|H|} = \frac{x\text{Re}H \frac{d\text{Re}H}{dx} + x\text{Im}H \frac{d\text{Im}H}{dx}}{|H|^2} \quad (6.33)$$

Recalling that

$$S_x^{\text{reH}} \triangleq \frac{x}{\text{Re}H} \frac{d\text{Re}H}{dx} \quad (6.34)$$

and

$$S_x^{\text{ImH}} \triangleq \frac{x}{\text{Im}H} \frac{d\text{Im}H}{dx}, \quad (6.35)$$

one can simplify (6.33) to

$$S_x^{|H|} = \frac{(ReH)^2 S_x^{ReH} + (ImH)^2 S_x^{ImH}}{|H|} \quad (6.36)$$

The sensitivity of the phase of the transfer function to changes in  $x$ ,  $S_x^\phi$ , is also easily obtained. By definition

$$\tan \phi = \frac{ImH}{ReH} \quad (6.37)$$

By implicit differentiation of (6.37) with respect to  $x$ , one obtains

$$\sec^2 \phi \frac{d\phi}{dx} = \frac{ReH \frac{dImH}{dx} - ImH \frac{dReH}{dx}}{(ReH)^2} \quad (6.38)$$

Since

$$S_x^\phi \triangleq \frac{x}{\phi} \frac{d\phi}{dx} \quad (6.39)$$

one obtains by substituting (6.38) into (6.39)

$$S_x^\phi = \frac{\cos^2 \phi}{\phi (ReH)^2} \left( xReH \frac{dImH}{dx} - xImH \frac{dReH}{dx} \right) \quad (6.40)$$

By use of the definitions in (6.34) and (6.35) and the relation

$$ReH = |H| \cos \phi, \quad (6.41)$$

the expression in (6.40) can be simplified to

$$S_x^\phi = \frac{1}{\phi |H|^2} (ReH \ ImH) \left( S_x^{ImH} - S_x^{ReH} \right) \quad (6.42)$$

Equations (6.36) and (6.42) are used in subroutine SENS to determine

$S_x^{|H|}$  and  $S_x^\phi$ , respectively.

This is possible since the quantities  $\phi$ ,  $|H|$ ,  $ReH$ , and  $ImH$  have been previously calculated and stored during generation of the Bode tables and

plots while  $S_x^{\text{Re}H}$  and  $S_x^{\text{Im}H}$  have been determined earlier in subroutine SENS.

#### VID DISCUSSION OF REVISED SENSITIVITY SUBROUTINE

In Appendix B is a revised version of subroutine SENSS which does the calculations of SENS and SENSS in a more efficient manner. This version of SENSS requires only 3/4 of the core storage required by the present SENS and SENSS and it utilizes a simpler algorithm that greatly reduces the number of mathematical operations required. This saves execution time and should increase the accuracy of the sensitivity calculations.

The version of SENSS given in Appendix B uses the same tagging procedure that is used in the present SENSS and calculates  $S_x^{H(j\omega)}$  with the use of equation (6.25) as does the present SENSS. However, the outputs of the present SENS, that is  $S_x^{|H|}$ ,  $S_x^\phi$ ,  $S_x^{\text{Re}H}$ , and  $S_x^{\text{Im}H}$  are related to the real and imaginary parts of  $S_x^{H(j\omega)}$ . Thus the rather complication sensitivity expressions now used in the present SENS and described in equations (6.23) through (6.42) are completely avoided.

Since, in general, the sensitivity expression  $S_x^{H(j\omega)}$  is a complex quantity, it can be written as

$$S_x^{H(j\omega)} = \text{Re} S_x^{H(j\omega)} + j \text{Im} S_x^{H(j\omega)} \quad (6.43)$$

But

$$H(j\omega) = |H(j\omega)| e^{j\phi} \quad (6.44)$$

where  $\phi$  is the phase of the transfer function  $H(j\omega)$ .

Thus one has

$$\begin{aligned} S_x^{H(j\omega)} &= S_x^{|H|} e^{j\phi} \\ &= S_x^{|H|} + S_x^{e^{j\phi}} \end{aligned} \quad (6.45)$$

Let us examine the rightmost term of (6.45) more closely. By definition

$$S_x^{e^{j\phi}} \triangleq \frac{dx}{e^{j\phi}} \frac{d}{dx} e^{j\phi}. \quad (6.46)$$

By use of the differentiation chain rule, the derivative expression in (6.46) can be simplified to

$$\frac{d}{dx} e^{j\phi} = j e^{j\phi} \frac{d\phi}{dx}. \quad (6.47)$$

Substituting (6.46) and (6.47) into (6.45) yields

$$S_x^{H(j\omega)} = S_x^{|H|} + jx \frac{d\phi}{dx}. \quad (6.48)$$

Using the definition given in (6.39) this equation can be rewritten as

$$S_x^{H(j\omega)} = S_x^{|H|} + j\phi S_x^\phi. \quad (6.49)$$

If the sensitivity parameter is a real quantity, then the expressions

$S_x^{|H|}$  and  $S_x^\phi$ , will also be real. Thus if (6.49) is compared with (6.43) and the real and imaginary parts equated (under the assumption that  $x$  is real) then one obtains

$$S_x^{|H|} = \operatorname{Re} S_x^{H(j\omega)} \quad (6.50)$$

and

$$S_x^\phi = \operatorname{Im} S_x^{H(j\omega)} \quad (6.51a)$$

or

$$S_x^\phi = \frac{1}{\phi} \operatorname{Im} S_x^{H(j\omega)} \quad (6.51b)$$

where  $x$  is a real variable.

The sensitivity of the real part of  $H(j\omega)$  to changes in  $x$ ,  $S_x^{\text{Re}H}$ , can also be obtained in terms of the real and imaginary parts of  $S_x^{H(j\omega)}$ . By definition

$$\text{Re}H(j\omega) = |H(j\omega)| \cos \phi \quad (6.52)$$

Thus

$$\begin{aligned} S_x^{\text{Re}H} &= S_x^{|H| \cos \phi} \\ &= S_x^{|H|} + S_x^{\cos \phi} \end{aligned} \quad (6.53)$$

But

$$\begin{aligned} S_x^{\cos \phi} &= \frac{x}{\cos \phi} \frac{d}{dx} \cos \phi \\ &= -x \tan \phi \frac{d\phi}{dx} \end{aligned} \quad (6.54)$$

Equation (6.54) can be rewritten as

$$S_x^{\cos \phi} = -\phi \tan \phi S_x^\phi \quad (6.55)$$

by use of the definition in (6.39). Thus equation (6.53) becomes, after the substitution of (6.55),

$$S_x^{\text{Re}H} = S_x^{|H|} - \phi \tan \phi S_x^\phi. \quad (6.56)$$

However, after (6.50), (6.51a) and (6.37) are substituted into (6.56), the expression becomes

$$S_x^{\text{Re}H} = \text{Re}S_x^{H(j\omega)} - \frac{\text{Im}H}{\text{Re}H} \text{Im}S_x^{H(j\omega)} \quad (6.57)$$

A similar expression can be derived for the sensitivity of the imaginary part of  $H(j\omega)$  with respect to changes in  $x$ ,  $S_x^{\text{Im}H}$ . By definition

$$\text{Im}H(j\omega) = |H(j\omega)| \sin \phi \quad (6.58)$$

Thus

$$\begin{aligned} S_x^{\text{Im}H} &= S_x^{|H| \sin \phi} \\ S_x^{|H|} + S_x^{\sin \phi} . \end{aligned} \quad (6.59)$$

By a similar mathematical technique, it can be shown that

$$S_x^{\sin \phi} = \frac{\phi}{\tan \phi} S_x^\phi \quad (6.60)$$

Substituting this expression into (6.59) gives

$$S_x^{\text{Im}H} = S_x^{|H|} + \frac{\phi}{\tan \phi} S_x^\phi . \quad (6.61)$$

which can be further simplified by the substitution of (6.50), (6.51a) and  
(6.37)

$$S_x^{\text{Im}H} = \text{Re} S_x^{H(j\omega)} + \frac{\text{Re} H}{\text{Im} H} \text{Im} S_x^{H(j\omega)} \quad (6.62)$$

Equations (6.50), (6.51b), (6.57) and (6.62) are used in the version of SENSS given in Appendix B. This is possible since the real quantities  $\phi$ ,  $\text{Re} H$ , and  $\text{Im} H$  have been calculated earlier in subroutine BODE while the complex quantity  $S_x^{H(j\omega)}$  is calculated in the Appendix B version of SENSS.

Note that (6.57) involves only 3 arithmetic operations while (6.25) involves 8 arithmetic operations plus the numerous operations involved in equations (6.23) and (6.26). The same comparison can be made between (6.62) and (6.28). With regard to  $S_x^{|H|}$ , (6.50) involves no arithmetic operations while (6.36) uses 8 operations. Similarly in determining  $S_x^\phi$  equation (6.51b) requires one arithmetic operation while (6.42) involves 6 operations. Since the sensitivity calculations must be redone for each frequency value, the number of arithmetic operations is quite substantially reduced with the version of SENSS given in Appendix B.

## VIE ROOT SENSITIVITY

The sensitivities of the poles and zeros of the transfer function to changes in some specified network parameter are also determined by NASAP by use of the tagging technique and the formulas given in [KU1, PA1] which will be summarized here.

Suppose we have a polynomial in the complex frequency variable  $s$ ,  $P(s)$ , which can also be expressed as

$$P(s) = \alpha(s) + x\beta(s) = \sum_{k=0}^N a_k s^k \quad (6.63)$$

where the degree of  $P(s)$  is  $N$  and  $P(s)$  has only positive powers of  $s$  and  $x$  is the sensitivity parameter. Furthermore the roots of the above polynomial are known, i.e.,  $N$  values of  $s$  are known such that

$$P(r_i) = 0 \quad i = 1, 2, \dots, N \quad (6.64)$$

where  $r_i$  is the  $i^{\text{th}}$  root of  $P(s)$ .

If the order of the root  $r_i$  is  $m$ , then the root sensitivity,  $s_x^{r_i}$ , as defined by

$$s_x^{r_i} \triangleq x \frac{dr_i}{ds} \quad (6.65)$$

can be expressed in terms of the given polynomials  $\alpha(s)$  and  $\beta(s)$

$$s_x^{r_i} = \left. \frac{\frac{d^{m-1}}{ds^{m-1}}(x\beta(s))}{\frac{d^m}{ds^m}(\alpha(s) + x\beta(s))} \right|_{s=r_i} \quad i=1, 2, \dots, N \quad (6.66)$$

If  $r_i$  is a simple root of  $P(s)$ , then

$$s_x^{r_i} = \left. \frac{-x\beta(r_i)}{\frac{d}{ds}(\alpha(s) + x\beta(s))} \right|_{s=r_i} \quad (6.67)$$

As was shown above, the transfer function  $H(s)$  can be written as

$$H(s) = \frac{N(s)}{D_1(s)} = \frac{A(s) + xB(s)}{C(s) + xD(s)} \quad (6.68)$$

where  $x$  is the sensitivity parameter.

If  $Z_i$  is a simple zero of  $H(s)$ , then the zero sensitivity with respect to  $x$  can be expressed as

$$s_x^{Z_i} = \left. \frac{xB(Z_i)}{\frac{d}{ds}(A(s) + xB(s))} \right|_{s=Z_i} = \left. \frac{-xB(Z_i)}{\frac{d}{ds} N(s)} \right|_{s=Z_i} \quad i=1, 2, \dots, M \quad (6.69)$$

where  $M$  is the number of distinct zeros and  $A(s)$ ,  $B(s)$ , and  $N(s)$  are defined in (6.23).

Similarly, if  $p_i$  is a simple pole of  $H(s)$ , then the pole sensitivity with respect to  $x$  can be written as

$$s_x^{p_i} = \left. \frac{xD(p_i)}{\frac{d}{ds}(C(s) + xD(s))} \right|_{s=p_i} = \left. \frac{-xD(p_i)}{\frac{d}{ds} D_1(s)} \right|_{s=p_i} \quad i=1, 2, \dots, N \quad (6.70)$$

where  $N$  is the number of distinct poles of  $H(s)$  and  $C(s)$ ,  $D(s)$ , and  $D_1(s)$  are defined in (6.68).

Since the polynomials  $A(s)$ ,  $xB(s)$ ,  $C(s)$ , and  $xD(s)$  have been determined by the tagging process during the evaluation of the loops in subroutines FLGRPH and HIGORL, the sensitivities of the poles and zeros are easily obtained by differentiating the denominator and numerator polynomials respectively and by evaluating the resultant polynomial and the appropriate tagged polynomial at the given pole or zero. These calculations are performed in subroutine ROOTSS.

## VIF EXAMPLES

In the foregoing sections of this chapter we have indicated many possibilities for sensitivity analysis with the aid of NASAP. We shall illustrate a few of these as follows:

1. Taking advantage of the background developed in Chapters IV and V on the unity feedback control system with lead cascade compensation see Fig. 6.1a (also Figs. 3.16 and 4.7) we obtain the sensitivity  $S_x^H$  of the transfer function  $VV4/VV5$  with respect to resistor  $R_1$ . This is a judicious choice of circuit element since one of the time constants of the control system plant is

$$\tau_1 = (R_1)(C_1)$$

Hence if  $C_1$  is constant in value the sensitivity determined for  $R_1$  is the same as that for the time constant  $\tau_1$ . The NASAP printout for this example is shown in Fig. 6.1. The corresponding zero and pole sensitivities are given in Fig. 6.2.

2. The second example Fig. 6.2 follows up with the uncompensated unity feedback control system for which we obtain the ramp response in Fig. 6.3 and the sensitivity function  $S_K^H$ . Thus we seek the sensitivity of the transfer function  $VV4/VV5$  with respect to system gain  $K$ , in this case  $K = 0.83$ . The NASAP printout is shown in Fig. 6.4 with the corresponding zero and pole sensitivities given in Fig. 6.5.

3. The third example furnishes sensitivity data, specifically  $S_K^H$  for the multiloop feedback compensated control system with  $K = 38$ . Since the pertinent NASAP model and transfer function print out were given in Fig. 5, they are not repeated here. The corresponding gain sensitivity print outs are given in Fig. 6.6 and the zero-pole sensitivity in Fig. 6.7.

4. For the final example we use one of the control subsystem of the Ranger space vehicle taken from Dorf [DOL, Problem 4.5]. The mission in 1965 was to scan the lunar surface with TV and other sensors. The requirement on the altitude control subsystem was to stabilize and control the Ranger spacecraft from second stage separation to lunar impact. Briefly a high gain antenna furnished input signals to the earth horizon sensor which in turn fed the gyro loop. The latter consisted of the spacecraft, as the plant, with the altitude gyro in cascade as shown in Fig. 6.8z along with its NASAP model. To give some feel for the cut and try process in determining the gyro loop gain K that will keep the step response overshoot under the required 5%, we show for three values of gain the print outs of  $S_K^H$ , zero-pole sensitivities  $S_H^{P_i}$  and the corresponding step response. For convenience in examining the print outs we tabulate the pertinent figure numbers.

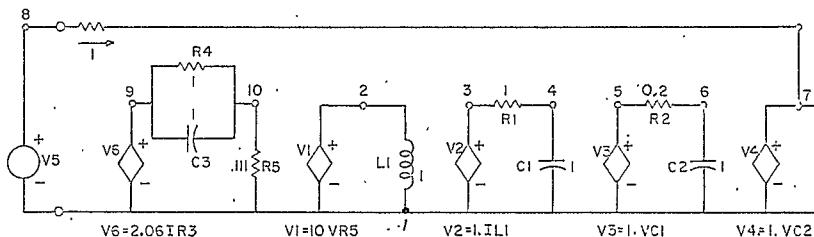
K	step response	$S_K^H$	$S_H^{P_i}$
6	6.8	Fig. 6.9	6.10
20	6.11	Fig. 6.12	6.13
75	6.14	Fig. 6.15	6.16

NASAP OP BLF LFAP - C: PERIODIC

8/20/69

HERTZ  
STEP RESISTORS

V1 1 2 1.. V1.5  
L1 2 1 1..  
V2 1 3 0 1..  
R1 3 4 1..  
C1 1 1..  
V3 1 5 1.0 1.. V1  
R2 5 2 1.. 2  
C2 6 1 1.. F  
V4 3 7 1.. V1.2  
R3 3 7 1.0..  
V5 1 9 1.0..  
V6 1 9 2.06 IR3  
C3 9 10 1.. F  
R4 9 10 1.. 3.. 0  
0.5 10 1.. 0.5 1..  
OUTPUT  
V1=V2=V3=1..  
FREQ 1.. 1.. 0.05  
TIME 0..  
EXECUT



TRANSFER FUNCTION V1/V5/V2

NUMBER OF LOOPS PER ORDER

( 1.00E 00 +1.00E 00 S )

1= 8  
2= 5  
3= 2

H(S)= 1.+33F 024-----

( 1.02E 02 +1.53E 02 S +6.51E 01 S +1.60E 01 S +1.00E 00 S )

ZERO OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

1 ->1.000E 01 0.0000E 00

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

1 ->6.794E 01 -0.24072E +1

2 ->5.797E 01 -0.24072E +1

3 ->0.300E 01 -1.48345E +1

4 ->0.1141E 02 0.80312E -09

## SENSITIVITY ANALYSIS

1	1.00E+00	0.00	0.00000000E+00	0.00000000E+00	0.00000000E+00	0.00000000E+00
2	-0.4949357E-00	0.00	0.1127019E-00	0.2023157E-00	0.4321705E-02	-0.6939700E-00
3	-0.3059327E-00	0.00	0.1245193E-00	0.24658671E-00	0.6701225E-02	-0.1029293E-00
4	-0.4949357E-00	0.00	0.1414529E-00	0.29774979E-00	0.652073995E-02	-0.2628980E-00
5	-0.1925203E-00	0.00	0.1774240E-00	0.3265453E-00	0.6211030E-02	-0.3625204E-00
6	-0.4949357E-00	0.00	0.1925203E-00	0.30204335E-00	0.6763109E-02	-0.4921432E-00
7	-0.5350000E-00	0.00	0.2235741E-00	0.6023446E-00	0.3565115E-02	-0.2192930E-00
8	-0.5350000E-00	0.00	0.4511494E-00	0.7313631E-00	0.8000124E-02	-0.1844244E-00
9	-0.5350000E-00	0.00	0.2813492E-00	0.64916161E-00	0.8705607E-02	-0.8053761E-01
10	-0.5350000E-00	0.00	0.3151277E-00	0.64916161E-00	0.8705607E-02	-0.1743555E-01
11	-0.4800E-02	0.00	0.335E-1131E-00	0.10624481E-01	0.11496693E-03	0.3461202E-01
12	-0.4800E-02	0.00	0.445E-1131E-00	0.1324509E-01	0.1224877E-03	0.12551542E-01
13	-0.3500E-02	0.00	0.445E-1131E-00	0.1324509E-01	0.1224877E-03	0.1207088E-00
14	-0.3500E-02	0.00	0.500E-1131E-00	0.1324509E-01	0.1224877E-03	0.1512649E-00
15	-0.5000E-02	0.00	0.5673449E-02	0.14654884E-01	0.1495346E-03	0.1762337E-00
16	-0.2000E-02	0.00	0.5673449E-02	0.14654884E-01	0.15016304E-03	0.1424545E-00
17	-0.156E-02	0.00	0.7074740E-02	0.1345778E-01	0.1569545E-03	0.1298194E-00
18	-0.156E-02	0.00	0.7991494E-02	0.1216161E-01	0.1669227E-03	0.9400636E-01
19	-0.5000E-02	0.00	0.9991301E-02	0.1158194E-01	0.1595935E-04	0.1626938E-01
20	-0.495E-02	0.00	0.1112419E-01	0.1115671E-01	0.17119967E-03	0.6020493E-01
21	-0.495E-02	0.00	0.125E-1125E-01	0.1117272E-01	0.1713202E-03	0.4331645E-01
22	0.1495E-01	0.00	0.1414529E-01	0.1054138E-01	0.17469115E-03	0.3492433E-01
23	0.2057492E-01	0.00	0.154E-1131E-01	0.1021616E-01	0.1720156E-03	0.4725487E-01
24	0.1495E-01	0.00	0.1774240E-01	0.1052050E-01	0.17611135E-03	0.16555677E-01
25	0.1925203E-01	0.00	0.1925203E-01	0.1040308E-01	0.1755166E-03	0.11410363E-01
26	0.1495E-01	0.00	0.2421149E-01	0.1094116E-01	0.1768494E-03	0.9072446E-02
27	0.2000E-01	0.00	0.24511531E-01	0.1040308E-01	0.1711537E-03	0.6154510E-02
28	0.4495335E-00	0.00	0.2813492E-01	0.1054456E-01	0.17469115E-03	0.4403248E-02
29	0.1997452E-00	0.00	0.3142439E-01	0.10241148E-01	0.17102123E-03	0.120724243E-02
30	0.14949357E-00	0.00	0.335E-1131E-01	0.1079494E-01	0.17171813E-03	0.1115548E-02
31	0.5350000E-00	0.00	0.391E-1131E-01	0.10324468E-01	0.17171813E-03	0.1075279E-02
32	0.6499371E-00	0.00	0.446E-1131E-01	0.1015193E-01	0.1794749E-03	0.10225046E-03
33	0.2000E-00	0.00	0.5623839E-01	0.1015005E-01	0.1784844E-03	0.214645305E-03
34	0.1499371E-00	0.00	0.5623839E-01	0.1015005E-01	0.1794749E-03	0.21614575E-03
35	0.7992405E-00	0.00	0.6107524E-01	0.1010440E-01	0.17804166E-03	0.11205241E-02
36	0.8497172E-00	0.00	0.6107524E-01	0.1010440E-01	0.17804166E-03	0.15094408E-04
37	0.3597418E-00	0.00	0.7945247E-01	0.10002036E-01	0.1780945E-03	0.1573438E-04
38	0.9495335E-00	0.00	0.8912474E-01	0.999917E-00	0.1790068E-03	-0.3399164E-05
39	0.4995335E-00	0.00	0.9995335E-01	0.9995335E-00	0.1791110E-03	-0.1233142E-04

			SLR(LRL(H))	SLR(S(1)(H))	SLR(S(A,S(1)))	SLR(S(T,H))
	-0.999998e-00	0.100000918e-00	-0.944352e-01	0.4542611e-00	0.1263394e-00	0.3367962e-00
	-0.999997e-00	0.11222191e-00	-0.969374e-01	-0.203315e-00	0.147453e-00	0.3952041e-00
	-0.999996e-00	0.12500000e-00	0.200762e-01	0.680191e-00	0.1661112e-00	0.4562461e-00
	-0.999995e-00	0.13710000e-00	0.161511e-01	0.761137e-00	0.18121234e-01	0.5245201e-00
	-0.999994e-00	0.14899999e-00	0.128511e-01	0.825876e-00	0.19512101e-01	0.5911462e-00
	-0.999993e-00	0.16070000e-00	0.100000e-01	0.882231e-00	0.2097594e-00	0.6687591e-00
	-0.999992e-00	0.17140000e-00	0.816671e-01	0.9098601e-00	0.1939422e-00	0.7215140e-00
	-0.999991e-00	0.18199999e-00	0.6667102e-01	0.9231833e-00	0.1701567e-00	0.7769867e-00
	-0.999990e-00	0.19233000e-00	0.5223311e-01	0.9348193e-00	0.1423604e-00	0.8268535e-00
	-0.999989e-00	0.202511e-00	0.4076993e-01	0.97320065e-01	0.1236221e-01	0.8475117e-00
	-0.999988e-00	0.212600e-00	0.3118654e-01	0.9900000e-01	0.1046221e-01	0.8682117e-00
	-0.999987e-00	0.222627e-00	0.24013945e-01	0.9934593e-00	-0.8122215e-01	0.8811109e-00
	-0.999986e-00	0.232643e-00	0.1864133e-01	0.98481032e-01	-0.2947882e-00	0.8929603e-00
	-0.999985e-00	0.242659e-00	0.1491670e-01	0.9747678e-01	-0.4734523e-00	0.7548524e-00
	-0.999984e-00	0.252675e-00	0.1166223e-01	0.9791225e-01	-0.7144049e-00	0.6538234e-00
	-0.999983e-00	0.262691e-00	0.8501164e-01	0.9818793e-01	-0.1079533e-00	0.53616337e-00
	-0.999982e-00	0.272707e-00	0.6262615e-01	0.9856801e-01	-0.1416242e-01	0.4051384e-00
	-0.999981e-00	0.282723e-00	0.4303668e-01	0.9862004e-01	-0.1823919e-01	0.29460956e-00
	-0.999980e-00	0.292739e-00	0.2982638e-01	0.9899460e-01	-0.2347249e-01	0.2014626e-00
	-0.999979e-00	0.302755e-00	0.1905759e-01	0.9981637e-01	-0.1234487e-01	0.1446988e-00
	-0.999978e-00	0.312771e-00	0.1115215e-01	0.2224243e-01	-0.1012479e-01	0.1010321e-00
	-0.999977e-00	0.322787e-00	0.5961664e-01	0.114332e-01	-0.2758496e-01	0.11724247e-01
	-0.999976e-00	0.332803e-00	0.1122411e-01	0.1126615e-01	-0.43492742e-01	0.1372284e-01
	-0.999975e-00	0.342819e-00	0.599579e-01	0.1259274e-01	-0.47380998e-01	0.1105636e-01
	-0.999974e-00	0.352835e-00	0.1141526e-01	0.1311275e-01	-0.65647648e-01	-0.10790959e-01
	-0.999973e-00	0.362851e-00	0.5998706e-01	0.14070708e-01	-0.82402433e-01	-0.10577784e-01
	-0.999972e-00	0.372867e-00	0.1172231e-01	0.15077046e-01	-0.98513544e-01	-0.10577784e-01
	-0.999971e-00	0.382883e-00	0.5999456e-01	0.1772231e-01	-0.10746262e-01	-0.10577784e-01
	-0.999970e-00	0.392899e-00	0.1923266e-01	0.1056748e-01	-0.92462811e-01	-0.1028767e-01
	-0.999969e-00	0.402915e-00	0.5999151e-01	0.175931e-01	-0.9442107e-01	-0.10194066e-01
	-0.999968e-00	0.412931e-00	0.1999151e-01	0.1056748e-01	-0.9549750e-01	-0.1013941e-01
	-0.999967e-00	0.422947e-00	0.5999151e-01	0.1804151e-01	-0.9643079e-01	-0.1010452e-01
	-0.999966e-00	0.432963e-00	0.1999151e-01	0.1056748e-01	-0.9697254e-01	-0.1005313e-01
	-0.999965e-00	0.442979e-00	0.5999151e-01	0.1805114e-01	-0.9749365e-01	-0.1003221e-01
	-0.999964e-00	0.452995e-00	0.1999151e-01	0.1056748e-01	-0.9801254e-01	-0.1001155e-01
	-0.999963e-00	0.462911e-00	0.5999151e-01	0.1805114e-01	-0.9853165e-01	-0.10000956e-01
	-0.999962e-00	0.472927e-00	0.1999151e-01	0.1056748e-01	-0.9905075e-01	-0.10000956e-01
	-0.999961e-00	0.482943e-00	0.5999151e-01	0.1804151e-01	-0.9956985e-01	-0.10000956e-01
	-0.999960e-00	0.492959e-00	0.1999151e-01	0.1056748e-01	-0.9998895e-01	-0.10000956e-01
	-0.999959e-00	0.502975e-00	0.5999151e-01	0.1802151e-01	-0.9999489e-01	-0.8562644e-02

(LOG(LK54))	Bx(0)	L0G(LK54) S(R50(1))	L0G(LK54) S(LK54(1))	L0G(LK54) S(LK54(1))	L0G(LK54) S(LK54(1))
-0.9999996e 00	0.1000001e 00	-0.1023486e 01	-0.3426944e 00	-0.8914692e 00	-0.4726326e 00
-0.4495972e 01	0.1120011e 01	-0.1114303e 01	-0.2751342e 00	-0.4112446e 00	-0.4031785e 00
-0.1999979e 02	0.1259261e 00	-0.1036357e 01	-0.2171716e 00	-0.1967261e 00	-0.3386912e 00
-0.4495985e 01	0.1115401e 01	-0.1115401e 01	-0.1115401e 00	-0.1115401e 00	-0.1115401e 00
-0.1999979e 00	0.1259261e 00	-0.1036357e 01	-0.2171716e 00	-0.1967261e 00	-0.3386912e 00
-0.7499985e 00	0.1171280e 00	-0.1301819e 01	-0.1301819e 00	-0.1301819e 00	-0.1301819e 00
-0.6999999e 00	0.1995263e 00	-0.7730377e 00	-0.9162211e-01	-0.7123278e 00	-0.1417552e 00
-0.6500030e 00	0.2231721e 00	-0.4497472e 00	-0.9792366e-01	-0.7626221e 00	-0.1095864e 00
-0.5900060e -01	0.2911986e 00	-0.6941176e 00	-0.1346792e 00	-0.5082740e 00	-0.8513631e-01
-0.5500070e -01	0.2711482e 00	-0.7545525e 00	-0.2112224e 00	-0.1470462e-01	-0.7182379e-01
-0.5000080e 00	0.3167277e 00	0.3158841e 00	-0.4252220e 00	-0.1090287e 01	-0.7052439e-01
-0.4500062e -01	0.3545134e 00	0.3604781e 00	0.1348401e 01	-0.5939570e-01	-0.4567780e-01
-0.4000064e 00	0.3981070e 00	0.1146370e 01	-0.4191942e 00	-0.3464212e 00	-0.1221363e 00
-0.3500066e 00	0.4665234e 00	0.8916667e 00	-0.6681704e 00	-0.1460561e 00	-0.1849393e 00
-0.3000068e 00	0.2011608e 00	0.4716054e 00	0.1036696e 00	-0.2817873e-01	-0.2520518e 00
-0.2500070e 00	0.5524628e 00	0.4622852e 00	0.1192703e 00	-0.3889353e-01	-0.2923964e 00
-0.2000074e 00	0.3709166e 01	-0.3043354e 00	-0.2744956e 00	-0.1061504e-01	-0.5304505e 00
-0.1500078e 00	0.1172165e 00	-0.5856643e-01	0.3011121e-01	-0.5265771e-01	-0.6117912e-01
-0.1000082e 00	0.7919327e 00	0.1212903e-01	0.3184605e 00	-0.9146633e-01	-0.6389352e 00
-0.5000085e 00	0.4912625e 00	0.5111972e-01	0.4611822e-01	-0.7241794e-01	-0.9955446e-00
-0.5901464e-01	0.9999986e 00	0.5617476e-01	0.4404104e 00	-0.5905324e-01	-0.1146424e 01
-0.8001464e-01	0.1111518e-01	0.56123411e-01	0.4632274e-01	-0.1247525e-01	-0.1247328e-01
-0.5999985e-01	0.12457428e 01	0.4990835e-01	-0.13449313e 01	-0.4412030e-01	-0.1302295e 01
-0.1499393e 00	0.1416124e 00	0.6264866e-01	-0.1849116e 00	-0.2405117e-01	-0.1462824e 01
-0.1999985e 01	0.1516890e 01	0.5616890e-01	-0.897778e-01	-0.4397028e 01	-0.1529036e 01
-0.4299255e 00	0.1117123e 01	0.4125713e-01	-0.6296392e-01	-0.1755113e-01	-0.1981835e 01
-0.2999386e 00	0.1994256e 01	0.77622734e 01	-0.3405390e-01	0.1231687e-01	-0.1634056e 01
-0.7499194e 00	0.4211616e 01	0.4521212e-01	-0.2549170e-01	-0.322294e-02	-0.1666794e 01
-0.4499285e 00	0.2811472e 01	0.3156667e-01	-0.1023628e-01	-0.1624118e-02	-0.1710646e 01
-0.4999985e 00	0.31672676e 01	0.7317392e-01	-0.1335124e-01	0.2301262e 02	-0.1735164e 01
-0.5499985e 00	0.4561123e 01	0.1335071e-01	-0.1023628e-01	0.1966242e-02	-0.1606442e 01
-0.5999985e 00	0.4911616e 01	0.5713232e-01	-0.9092376e-02	0.3064786e-03	-0.1767080e 01
-0.6499985e 00	0.5444414e 01	0.4716132e-01	-0.1476356e-02	0.4321846e-03	-0.1815533e 01
-0.6999985e 00	0.5301152e 01	0.2474949e-01	-0.6111353e-02	0.2024846e-03	-0.1845967e 01
-0.7495525e-01	0.3622424e 01	0.5256018e-01	-0.4302553e-02	0.50607712e-04	-0.1878607e 01
-0.7999984e 00	0.6407234e 01	0.5559717e-01	-0.4042260e-02	0.9186135e-05	-0.1912766e 01
-0.8499279e 00	0.110464238e 01	0.7699272e-01	-0.4202629e-02	-0.4130590e-04	-0.1942995e 01
-0.8999984e 00	0.77662476e 01	0.6444301e-01	-0.2659272e-02	-0.6717151e-04	-0.1986667e 01
-0.9499984e 00	0.3912276e 01	0.2674571e-01	-0.2113624e-02	-0.4686258e-04	-0.2025816e 01
-0.9999985e 00	0.9999969e 01	0.2657759e-01	-0.1696406e-02	-0.6962588e-04	-0.2066379e 01

LUG(SENSE(0)) 3347458714  
LUG(SENSE(1)) +++++++  
LUG(SENSE(LABS(0))) 000006.0000

—LOG(FREQ)



## SENSITIVITY OF ZEROS AND POLES OF TRANSFER FUNCTION

ZERO	REAL	IMAG	REAL	SENSITIVITY
1	-0.100000E-01	0.000000E+00	0.915500E-06	0.000000E+00

POLE	REAL	IMAG	REAL	SENSITIVITY
1	-0.179679E-01	0.2407153E-01	0.1949336E+00	-0.2548719E-01
2	-0.179679E-01	-0.4407153E-01	0.1949336E+00	0.2548719E-01
3	-0.100000E-01	-0.4433796E-11	-0.5181970E+00	0.4601166E-07
4	-0.1141942E-02	0.8030199E-09	0.1148264E+01	-0.5163200E-05

SII

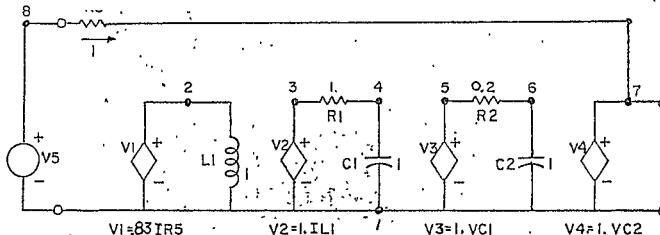
7/14/69

--- NASA-P-PROBLEM .UNCOMPENSATED. PLANT.

--- NSME HEAT%  
--- RAMP RES-ONSE

V1	1	2	C.B3	R3
L1	2	1	1.0	I11
V2	1	3	1.0	I11
R1	3	4	1.0	
C1	4	1	1.0	
V3	1	5	1.0	Vc1
R2	5	6	0.2	
C2	6	1	1.0	
V4	1	7	1.0	Vc2
R3	8	7	1.0	
V5	1	8	1.0	

OUTPUT  
VV4/VV5/1  
FREQ 0.0 1.0 0.05  
TIME 1/5.  
EXECUTE



--- TRANSFER FUNCTION VV4/VV5/V1

NUMBER OF LOOPS PER ORDER

1= 4

2= -1

( -1.00E-00 )

--- H(S) = -4.150E 00\*

( 4.15E 00 +5.00E 00 S +6.00E 00 S +1.00E 00 S )

--- ZERO OF TRANSFER FUNCTION

--- NONE

--- POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

1	-0.40461E 00	0.79736E 00
2	-0.40461E 00	-0.79736E 00
3	-0.51900E 01	0.62528E-12

## RAMP RESPONSE FUNCTION

F(T) =

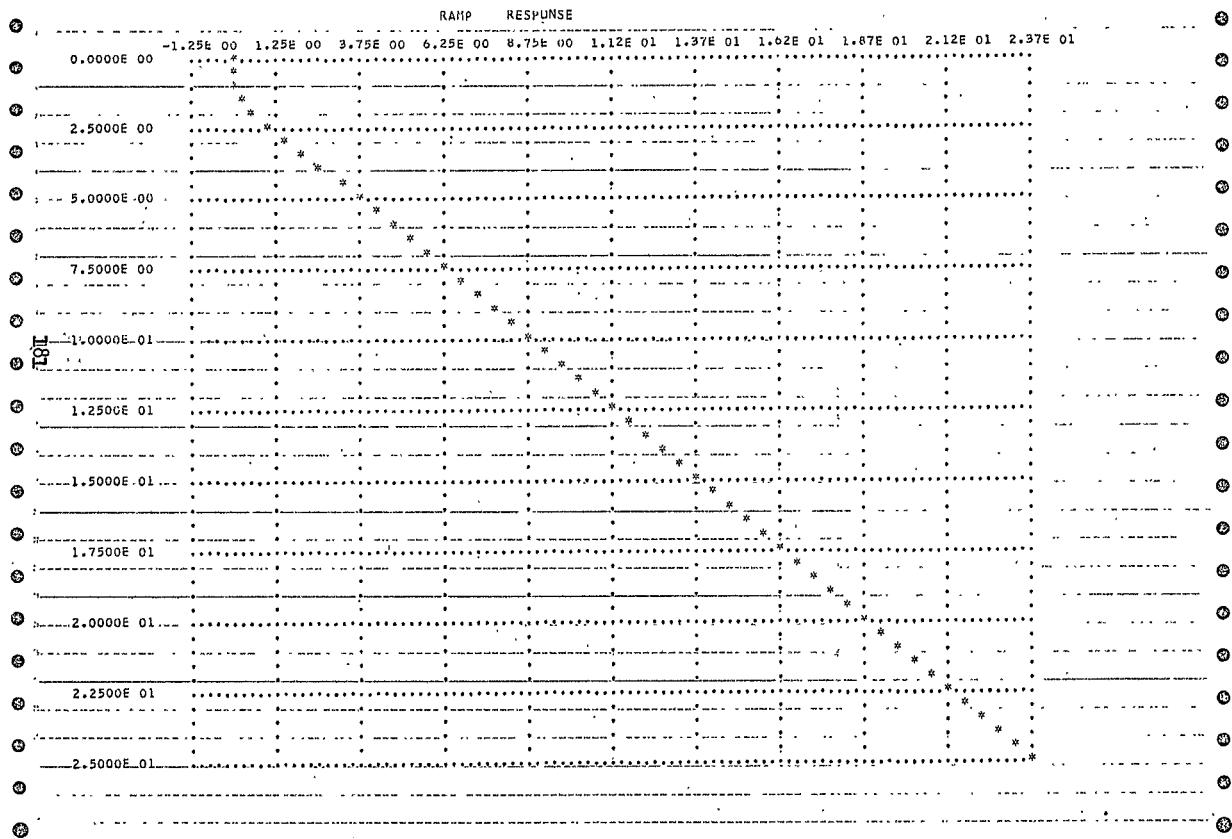
(0.5991E 00 J 0.3017E 00 ) E (-0.4046E 00 J 0.7974E 00 ) T  
 (-0.4046E 00 J=0.7974E 00 ) T  
 (-0.5991E 00 J-0.3017E 00 ) E  
 (-0.5191E 01 J 0.6253E-12 ) T  
 (-0.6542E-02 J=0.1134E-12 ) E  
 (-0.6024E 00 J-0.5000E 03 ) E  
 (-0.0000E 00 J 0.1000E-02 ) T  
 (-0.6024E 00 J 0.5000E 03 ) E

180

## RAMP RESPONSE

TIME	VV4/VV5
0.0000E 00	-0.18350231E-04
0.5000E 00	0.6328405E-02
0.1000E 01	0.65632403E-01
0.1500E 01	0.22013344E 00
0.2000E 01	0.51319776E 00
0.2500E 01	0.91621377E 00
0.3000E 01	0.14124557E 01
0.3500E 01	0.1971751E 01
0.4000E 01	0.25636339E 01
0.4500E 01	0.3102348E 01
0.5000E 01	0.37496081E 01
0.5500E 01	0.43153048E 01
0.6000E 01	0.48558273E 01
0.6500E 01	0.53730450E 01
0.7000E 01	0.58193206E 01
0.7500E 01	0.63587666E 01
0.8000E 01	0.68396826E 01
0.8500E 01	0.73197384E 01
0.9000E 01	0.78024302E 01
0.9500E 01	0.8289810E 01
1.0000E 02	0.87820425E 01
0.1050E 02	0.92190327E 01
0.1100E 02	0.977195935E 01
0.1150E 02	0.10282449E 02
0.1200E 02	0.10706363E 02
0.1250E 02	0.11290394E 02
0.1300E 02	0.1179327E 02
0.1350E 02	0.12296152E 02
0.1400E 02	0.12797535E 02
0.1450E 02	0.1329794E 02
0.1500E 02	0.13797764E 02
0.1550E 02	0.14297008E 02
0.1600E 02	0.14796228E 02
0.1650E 02	0.152959361E 02
0.1700E 02	0.15794614E 02
0.1750E 02	0.16294052E 02
0.1800E 02	0.16793701E 02
0.1850E 02	0.17293533E 02
0.1900E 02	0.17793668E 02
0.1950E 02	0.182935339E 02
0.2000E 02	0.18793594E 02
0.2050E 02	0.19293655E 02
0.2100E 02	0.19793671E 02
0.2150E 02	0.20293640E 02
0.2200E 02	0.20793579E 02
0.2250E 02	0.21293457E 02
0.2300E 02	0.21793320L 02
0.2350E 02	0.22293152E 02
0.2400E 02	0.22792984E 02
0.2450E 02	0.23292801E 02
0.2500E 02	0.23792618E 02

A69380



LGD(FREQ)	FREQ	ABS(SENS(H))	PHI(SENS(H))	LGD/ABS(SENS(H))
0.0000E+00	0.1000000E+01	0.1009591E+01	0.1794444E+02	0.4145510E-02
0.4999995E-01	0.1122018E+01	0.1008610E+01	0.1795293E+02	0.2834388E-02
0.9999990E-01	0.1242225E+01	0.1007447E+01	0.1795078E+02	0.1940213E-02
1.4999985E-01	0.1362432E+01	0.1006300E+01	0.1797606E+02	0.1304348E-02
1.9999980E-01	0.1482640E+01	0.1005199E+01	0.1798189E+02	0.8572450E-03
2.4999975E-01	0.1602848E+01	0.1004131E+01	0.1798646E+02	0.5720134E-03
2.9999970E-01	0.1723078E+01	0.1003080E+01	0.1798998E+02	0.3746671E-03
3.4999975E-01	0.1843296E+01	0.1002086E+01	0.1799265E+02	0.2430929E-03
3.9999966E-01	0.1963518E+01	0.1001056E+01	0.1799463E+02	0.1373582E-03
4.4999966E-01	0.2083730E+01	0.1000243E+01	0.1799611E+02	0.1010470E-03
4.9999956E-01	0.3162274E+01	0.1000149E+01	0.1799720E+02	0.6460656E-04
5.4999956E-01	0.3548130E+01	0.1000095E+01	0.1799798E+02	0.4141537E-04
5.9999946E-01	0.3981067E+01	0.1000061E+01	0.1799856E+02	0.2590642E-04
6.4999946E-01	0.4466829E+01	0.1000038E+01	0.1799897E+02	0.1536510E-04
6.9999938E-01	0.5011865E+01	0.1000024E+01	0.1799922E+02	0.1000025E-04
7.4999938E-01	0.5623405E+01	0.1000015E+01	0.1799958E+02	0.6626757E-05
7.9999925E-01	0.6209363E+01	0.1000006E+01	0.1799974E+02	0.3727562E-05
8.4999922E-01	0.6794942E+01	0.1000004E+01	0.1799981E+02	0.1656698E-05
8.9999918E-01	0.7379472E+01	0.1000001E+01	0.1799987E+02	0.4141752E-06
9.4999918E-01	0.8912471E+01	0.1000001E+01	0.1799991E+02	0.0000000E+00
9.9999905E-01	0.9999970E+01	0.1000000E+01	0.1799991E+02	0.0000000E+00

LGD(FREQ)	FREQ	LUG(ABS(SENS))	LUG(PHI(SENS))	LGD/ABS(SENS)	LGD/PHI(SENS)
-8.50E-04	-3.50E-04	1.50E-04	6.50E-04	1.15E-03	1.65E-03
-0.0000E+00	-0.5000E-01	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00
2.5000E-01	0.0000E+00	*	*	*	*
5.0000E-01	0.0000E+00	*	*	*	*
7.5000E-01	0.0000E+00	*	*	*	*
1.0000E-00	0.0000E+00	*	*	*	*

LGD(FREQ)	FREQ	PHI(SENS)	LGD/PHI(SENS)
-2.00E-02	-1.60E-02	-1.20E-02	-8.00E-02
-4.00E-01	0.00E+00	4.00E+01	8.00E+01
0.0000E+00	1.20E+02	1.60E+02	2.00E+02

LGD(FREQ)	FREQ	PHI(SENS)	LGD/PHI(SENS)
-2.5000E-01	0.0000E+00	*	*
-5.0000E-01	0.0000E+00	*	*
-7.5000E-01	0.0000E+00	*	*
-1.0000E-00	0.0000E+00	*	*

LGD(FREQ)	FREQ	PHI(SENS)	LGD/PHI(SENS)
-7.5000E-01	0.0000E+00	*	*
-5.0000E-01	0.0000E+00	*	*
-2.5000E-01	0.0000E+00	*	*
-1.0000E-00	0.0000E+00	*	*

SENSITIVITIES OF ZEROS AND POLES OF TRANSFER FUNCTION

ZERO	REAL	IMAG	REAL	SENSITIVITY
				IMAG
1	0.000000E 00	0.000000E 00	0.990000E 36	0.000000E 00

POLE	REAL	IMAG	REAL	SENSITIVITY
				IMAG
1	-0.4046128E 00	0.7973596E 00	-0.4830204E 00	0.3556389E 00
2	-0.4046128E 00	-0.7973596E 00	-0.4830204E 00	-0.3556389E 00
3	-0.5190775E 01	0.625276E-12	-0.3395369E-01	-0.2135725E-13

$$-0.339(-.519) = +.175$$

$$S_x^P = -0.175 = \frac{-x B(p)}{P(p)}$$

$$(-.48 + j .36)(-.40 + j .80)$$

where  $P(p) = A_{pp} \times B(p)$

$$= +(.48)(.40) - (.36)(.80) + j [(-.96)(.36) + (-.48)(.80)]$$

$$= \frac{-RSJ \times K_0}{(RDEN)^2}$$

$$= .192 - .288 - j [ .144 + .384 ]$$

$$= -.096 - j .528$$

$$\frac{R}{(1 + j)^2} K$$

$$S_x^P = .096 + j .528$$

## SENSITIVITY ANALYSIS \*\*

	LBC(SENS(H1))	RHT(SENS(H1))	LDC(LABS(SENS(H1)))
10	-0.9992795E-00	0.1000001E-00	0.1012636E-01
11	-6.9492797E-00	0.11123289E-01	0.11197491E-01
12	-0.89994997E-00	0.1258926E-00	0.1004095E-01
13	-0.84993982E-00	0.1412538E-00	0.1005084E-01
14	-0.79999980E-00	0.1584894E-00	0.10066290E-01
15	-0.74916655E-00	0.1770240E-00	0.1017515E-01
16	-0.69914996E-00	0.1955263E-00	0.1029556E-01
17	-0.65010005E-00	0.22334721E-00	0.1041587E-01
18	-0.60000000E-00	0.2511886E-00	0.1010426E-01
19	-0.5509161E-00	0.28113382E-00	0.1016841E-01
20	-0.50000001E-00	0.3162277E-00	0.1020043E-01
21	-0.45921635E-00	0.3544113E-00	0.1013515E-01
22	-0.40000002E-00	0.3981070E-00	0.1027440E-01
23	-0.35000005E-00	0.4466613E-00	0.1015228E-01
24	-0.3000003E-00	0.5011864E-00	0.1015746E-01
25	-0.25000002E-00	0.5623409E-00	0.1019252E-01
26	-0.20000000E-00	0.6309568E-00	0.1014089E-01
27	-0.16023255E-00	0.7026405E-00	0.1014272E-01
28	-0.10000000E-00	0.7943273E-00	0.10141489E-01
29	-0.80090055E-01	0.8912498E-00	0.10154576E-01
30	-0.5964664E-06	0.99999986E-00	0.10176161E-01
31	-0.4992664E-01	0.11220155E-01	0.10169220E-01
32	-0.4995333E-01	0.12556922E-01	0.10166630E-01
33	-0.4999252E-01	0.14125355E-01	0.10153201E-01
34	-0.1999592E-30	0.1584890E-01	0.1061799E-01
35	-0.24691085E-00	0.1772723E-01	0.1014561E-01
36	-0.29997868E-00	0.1995256E-01	0.10604497E-01
37	-0.3495558E-00	0.22330715E-01	0.1058920E-01
38	-0.3999493E-00	0.2511881E-01	0.10567772E-01
39	-0.4499643E-01	0.2811457E-01	0.1040209E-01
40	-0.4999585E-00	0.3162267E-01	0.1051006E-01
41	-0.5499985E-00	0.3544122E-01	0.1074770E-01
42	-0.5997449E-01	0.3981061E-01	0.10484803E-01
43	-0.6499881E-01	0.4466616E-01	0.1028124E-01
44	-0.6999983E-00	0.5011852E-01	0.1042934E-01
45	-0.7490465E-01	0.5623393E-01	0.10462923E-01
46	-0.7999986E-00	0.6309554E-01	0.1057462E-01
47	-0.8499579E-00	0.7079423E-01	0.1028557E-01
48	-0.89992981E-01	0.7943247E-01	0.1115892E-01
49	-0.9499883E-00	0.8512327E-01	0.11157927E-01
50	-0.9999985E-01	0.99999965E-01	0.1257830E-01
51	-0.10402292E-01	0.11220112E-02	0.1313234E-01
52	-0.1099998E-01	0.1258918E-02	0.11919462E-02
53	-0.11459988E-01	0.1421530E-02	0.18526573E-00
54	-0.11999998E-01	0.15084868E-02	0.15474832E-00
55	-0.12439925E-01	0.1779260E-02	0.2450007E-00
56	-0.1299997E-01	0.1995520E-02	0.2216687E-00
57	-0.1340558E-01	0.2228705E-02	0.145514E-00
58	-0.1390998E-01	0.2511873E-02	0.9719779E-01
59	-0.14405297E-01	0.2818362E-02	0.452293E-01
60	-0.1499997E-01	0.3162257E-02	0.4513240E-01
61	-0.15446525E-01	0.3568112E-02	0.2134272E-01
62	-0.15929949E-01	0.3981049E-02	0.2160375E-01
63	-0.16466232E-01	0.4466203E-02	0.1515725E-01
64	-0.1699997E-01	0.5011835E-02	0.1053014E-01
65	-0.17465927E-01	0.5623376E-02	0.2933274E-02
66	-0.1799997E-01	0.6309533E-02	0.918081E-02
67	-0.18494949E-01	0.7079402E-02	0.665115E-02
68	-0.1899997E-01	0.7943224E-02	0.2572703E-02
69	-0.1949997E-01	0.89112447E-02	0.1814594E-02
70	-0.1999997E-01	0.9999934E-02	0.1240908E-02

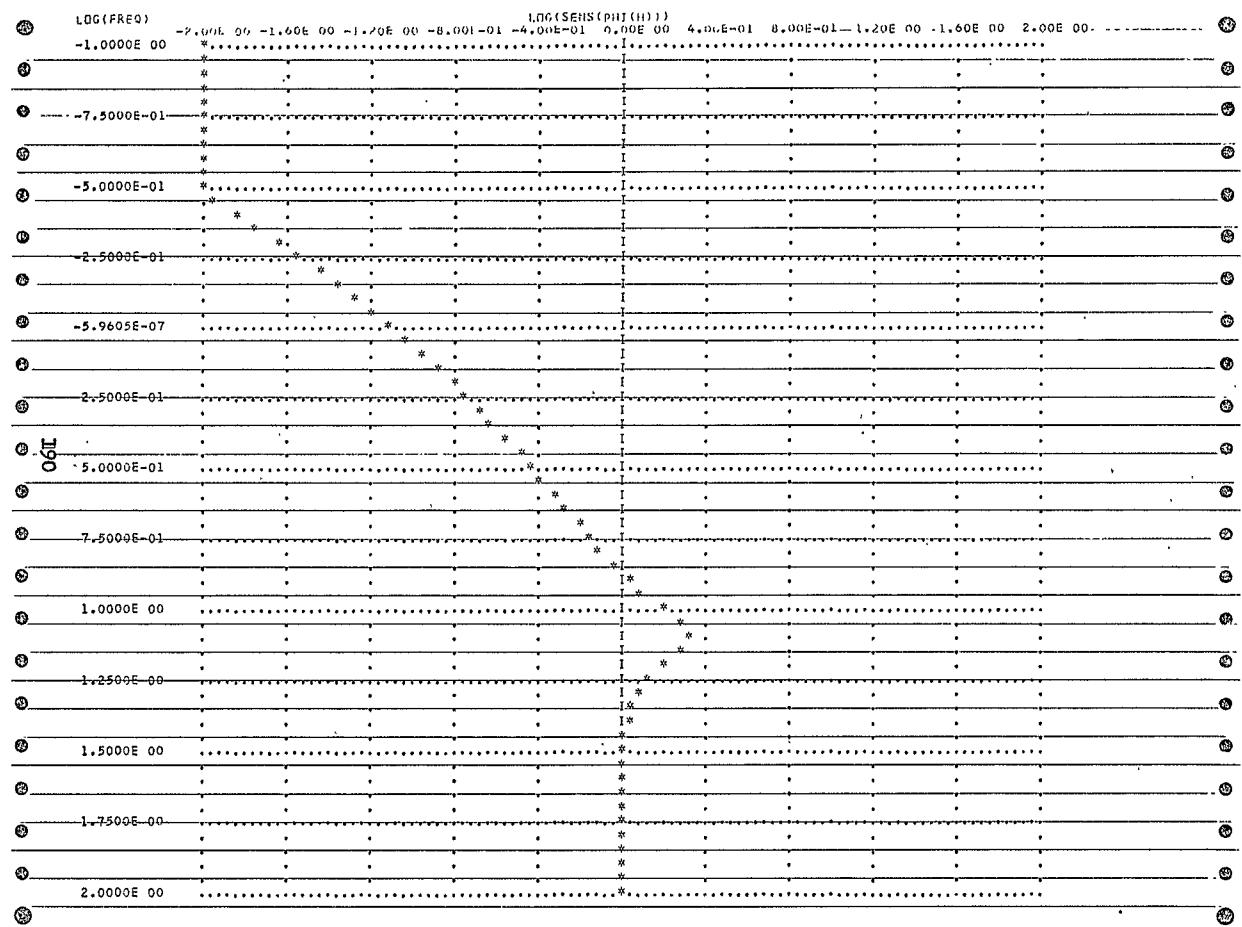
	LOG(L&EQ)	LUG(L&BS)(S&HS))
1	-2.37E 00	-2.12E 00 -1.87E 00 -1.62E 00 -1.37E 00 -1.12E 00 -8.75E-01 -6.25E-01 -3.75E-01 -.25E-01 1.25E-01
2	-1.0000E-00	*
3	*	*
4	*	*
5	*	*
6	-7.5000E-01	*
7	*	*
8	*	*
9	*	*
10	-5.0000E-01	*
11	*	*
12	*	*
13	-2.5000E-01	*
14	*	*
15	*	*
16	5.9605E-07	*
17	*	*
18	*	*
19	2.5000E-01	*
20	*	*
21	*	*
22	*	*
23	5.0000E-01	*
24	*	*
25	*	*
26	7.5000E-01	*
27	*	*
28	*	*
29	1.0000E-00	*
30	*	*
31	*	*
32	1.2500E 00	*
33	*	*
34	1.5000E 00	*
35	*	*
36	1.7500E 00	*
37	*	*
38	2.0000E-00	*



		F1(F2)	S1(M1)(M2)	S1(M1)(M2)	S1(M1)(M2)	S1(M1)(M2)	
①	-0.99994996E-00	0.1000001E-00	-0.9948553E-00	-0.1005266E-01	-0.1002832E-01	0.1931363E-02	
②	-C_9499997E-00	0.1122019E-00	-0.9991519E-00	-0.1000567E-01	-0.1003284E-01	0.1602272E-02	
③	-0.8992997E-00	0.1258926E-00	-0.9994920E-00	-0.1008174E-01	-0.1004086E-01	0.1862695E-02	
④	-0.8499995E-00	0.1412538E-00	-0.9998618E-00	-0.1010143E-01	-0.1005072E-01	0.2187994E-02	
⑤	-0.7992898E-00	0.1584824E-00	-0.1012546E-01	-0.1006272E-01	-0.1006272E-01	0.2550323E-02	
⑥	-0.7499999E-00	0.1778208E-00	-0.1000606E-01	-0.1015450E-01	-0.1007772E-01	0.3170911E-02	
⑦	-0.6996499E-00	0.1995203E-00	-0.1000870E-01	-0.1018930E-01	-0.1009469E-01	0.3789346E-02	
⑧	-0.6500000E-00	0.2238721E-00	-0.1000978E-01	-0.1023054E-01	-0.1011527E-01	0.4468482E-02	
⑨	-0.6000000E-00	0.2511886E-00	-0.1000959E-01	-0.1027870E-01	-0.1013939E-01	0.5759098E-02	
⑩	-0.5500001E-00	0.2818132E-00	-0.10009609E-01	-0.1033403E-01	-0.1016702E-01	0.7188428E-02	
⑪	-0.5000001E-00	0.3162277E-00	-0.1000956E-01	-0.1038432E-01	-0.1019818E-01	0.9024448E-02	
⑫	-0.4500002E-00	0.3548133E-00	-0.1000943E-01	-0.1046501E-01	-0.1023251E-01	0.1116992E-01	
⑬	-0.4000002E-00	0.3981070E-00	-0.1000979E-01	-0.1053856E-01	-0.1026928E-01	0.1434250E-01	
⑭	-0.3500003E-00	0.4466633E-00	-0.1000974E-01	-0.1061491E-01	-0.1030746E-01	0.1857306E-01	
⑮	-0.3000003E-00	0.5011868E-00	-0.1000947E-01	-0.1069412E-01	-0.1034562E-01	0.2220189E-01	
⑯	-0.2500004E-00	0.5623409E-00	-0.10009277E-01	-0.1076424E-01	-0.1038212E-01	0.2637518E-01	
⑰	-0.2000004E-00	0.6326556E-00	-0.10009142E-01	-0.1083020E-01	-0.1041501E-01	0.3524475E-01	
⑱	-0.1500005E-00	0.7079450E-00	-0.10009151E-01	-0.1080455E-01	-0.1044228E-01	0.4345018E-01	
⑲	-0.1000005E-00	0.7943233E-00	-0.10009150E-01	-0.1092379E-01	-0.1046183E-01	0.5315230E-01	
⑳	-0.5000005E-01	0.8912498E-00	-0.10009718E-01	-0.1094366E-01	-0.1047183E-01	0.6450325E-01	
㉑	-0.496E-0446E-00	0.9999986E-00	-0.10009521E-01	-0.1094018E-01	-0.1047009E-01	0.7746867E-01	
㉒	0.4999364E-01	0.1122015E-01	-0.10009249E-01	-0.1090918E-01	-0.1045459E-01	0.9282982E-01	
㉓	0.9999985E-01	0.125-932E-01	-0.10009153E-01	-0.1084620E-01	-0.1042310E-01	0.1111939E-00	
㉔	0.1499959E-00	0.1412538E-01	-0.10009176E-01	-0.1076412E-01	-0.1037300E-01	0.1299878E-00	
㉕	0.1999592E-00	0.1584820E-01	-0.10009165E-01	-0.1060303E-01	-0.1030151E-01	0.1524829E-00	
㉖	0.2499385E-00	0.1778273E-01	-0.10009147E-01	-0.1040948E-01	-0.1020493E-01	0.1779776E-00	
㉗	0.2999280E-00	0.1995256E-01	-0.10009156E-01	-0.1015816E-01	-0.1007908E-01	0.2048093E-00	
㉘	0.3499988E-00	0.2238715E-01	-0.10009173E-02	-0.9838155E-00	-0.9919078E-00	0.2373461E-00	
㉙	0.3999590E-00	0.2511881E-01	-0.10009132E-01	-0.9438252E-00	-0.9719261E-00	0.2758949E-00	
㉚	0.4499983E-01	0.2818732E-01	-0.10009142E-01	-0.8946664E-00	-0.9473332E-00	0.3171780E-00	
㉛	0.4999985E-00	0.3162625E-01	-0.10009207E-01	-0.8348843E-01	-0.9174420E-00	0.3633568E-00	
㉜	0.5499986E-00	0.3549812E-01	-0.10008703E-01	-0.7630451E-00	-0.8815222E-00	0.4150307E-00	
㉝	0.5999888E-00	0.3981061E-01	-0.10009242E-01	-0.6775975E-00	-0.8387988E-00	0.4727807E-00	
㉞	0.6499981E-00	0.4466816E-01	-0.10009173E-01	-0.5761818E-00	-0.7884094E-00	0.5373604E-00	
㉟	0.6999882E-00	0.5011852E-01	-0.10009264E-01	-0.4858189E-00	-0.7292592E-00	0.6098959E-00	
㉟	0.7499985E-00	0.5623393E-01	-0.10009154E-01	-0.3193306E-00	-0.5996650E-00	0.6972721E-00	
㉟	0.7999886E-00	0.6309556E-01	-0.10009221E-01	-0.15930323E-01	-0.5765160E-01	0.7876891E-00	
㉟	0.8499979E-01	0.7079423E-01	-0.10009256E-01	-0.5313903E-01	-0.4734304E-00	0.9011121E-00	
㉟	0.8999881E-01	0.7943247E-01	-0.10009248E-01	-0.3262816E-01	-0.3365894E-01	0.1050014E-01	
㉟	0.9499983E-00	0.8912474E-01	-0.10009171E-01	-0.7281504E-00	-0.1359242E-00	0.1250772E-01	
㉟	0.9999985E-00	0.9999956E-01	-0.10009116E-01	-0.1341416E-01	-0.1870804E-01	0.1537979E-01	
㉟	0.1049997E-01	0.1122012E-02	-0.8408953E-01	-0.2348030E-01	-0.6740179E-01	0.1902025E-01	
㉟	0.1093395E-01	0.1258918E-02	-0.8733910E-01	-0.3016895E-01	-0.1038445E-01	0.2092329E-01	
㉟	0.1149999E-01	0.1412530E-02	-0.8576619E-01	-0.2713143E-01	-0.8576546E-01	0.1885491E-01	
㉟	0.1196995E-01	0.158A886E-02	-0.8476266E-01	-0.2024966E-01	-0.5124903E-01	0.1520636E-01	
㉟	0.1249997E-01	0.1995256E-02	-0.7782662E-01	-0.2555774E-01	-0.1569996E-01	0.2849996E-01	0.1294746E-01
㉟	0.1293397E-01	0.1995256E-02	-0.1357151E-01	-0.13200767E-01	-0.1600046E-01	0.1166735E-01	
㉟	0.1349998E-01	0.2238708E-02	-0.8032620E-01	-0.1183975E-01	-0.9198093E-01	0.1095850E-01	
㉟	0.1392995E-01	0.2511873E-02	-0.4781847E-01	-0.1108058E-01	-0.5403042E-01	0.1056096E-01	
㉟	0.1449997E-01	0.2818362E-02	-0.2908163E-01	-0.1064562E-01	-0.3228283E-01	0.1033345E-01	
㉟	0.1499987E-01	0.3162257E-02	-0.1794142E-01	-0.1039672E-01	-0.1953793E-01	0.1020075E-01	
㉟	0.1549997E-01	0.3548112E-02	-0.11122295E-01	-0.1023878E-01	-0.1194000E-01	0.1012211E-01	
㉟	0.1592228E-01	0.3981049E-02	-0.6942270E-02	-0.1014702E-01	-0.7349968E-02	0.1007488E-01	
㉟	0.1649997E-01	0.4466803E-02	-0.4349393E-02	-0.1009069E-01	-0.4549026E-02	0.1004610E-01	
㉟	0.1692997E-01	0.5011838E-02	-0.2725046E-02	-0.1005658E-01	-0.2827644E-02	0.1002865E-01	
㉟	0.1749997E-01	0.5623376E-02	-0.1712024E-02	-0.1003515E-01	-0.1763334E-02	0.1001769E-01	
㉟	0.1799992E-01	0.6309535E-02	-0.1074257E-02	-0.1002202E-01	-0.1102448E-02	0.1001109E-01	
㉟	0.1849997E-01	0.7079940E-02	-0.6776147E-03	-0.1001378E-01	-0.6904602E-03	0.1000692E-01	
㉟	0.1894527E-01	0.7943224E-02	-0.4243536E-03	-0.1000850E-01	-0.4329681E-03	0.1000420E-01	
㉟	0.1949997E-01	0.8912474E-02	-0.26857970E-03	-0.1000515E-01	-0.2717972E-03	0.1000248E-01	
㉟	0.1998495E-01	0.9889934E-02	-0.1629864E-03	-0.1000355E-01	-0.1707077E-03	0.1000182E-01	

	INSTRUMENT	TYPE	LUG(SENS(AB)(1))	LUG(SENS(AB)(1))	LUG(SENS(AB)(1))	LUG(SENS(AB)(1))
④	-0.19900569E-00	0.10000001E-00	-0.04955542E-03	-0.2290251E-02	-0.141622E-02	-0.2865559E-01
④	-0.9499997E-00	0.1122019E-00	-0.3684676E-03	-0.2842668E-02	0.1423670E-02	-0.2795262E-01
④	-0.39999976E-00	-0.1256926E-00	-0.2214594E-03	-0.3935057E-02	-0.177124E-02	-0.2729857E-01
④	-0.8499998E-00	0.1412528E-00	-0.6093369E-04	-0.4382957E-02	0.219701E-02	-0.2659961E-01
④	-0.7999998E-00	0.1384494E-00	-0.1046441E-03	-0.5444737E-02	-0.1715584E-02	-0.2635306E-01
④	-0.7782000E-00	0.1177620E-00	-0.2624921E-03	-0.6652383E-02	-0.1341920E-02	-0.2505714E-01
④	-0.6999999E-00	0.1199526E-00	-0.3775638E-03	-0.8144933E-02	-0.4091356E-02	-0.2424435E-01
④	-0.6500000E-00	0.2238721E-00	-0.4020509E-03	-0.9896070E-02	-0.4977502E-02	-0.2332562E-01
④	-0.6000000E-00	0.2111488E-00	-0.2959464E-03	-0.1149382E-01	-0.6010149E-02	-0.7249645E-01
④	-0.5000000E-00	0.3813322E-00	-0.1616763E-03	-0.1467588E-01	-0.7193547E-02	-0.2143366E-01
④	-0.4999999E-00	0.3414224E-00	-0.1093745E-03	-0.1498154E-01	-0.4522201E-02	-0.2044577E-01
④	-0.4500002E-00	0.3548133E-00	-0.2661970E-02	-0.1973971E-01	-0.9981997E-02	-0.1944242E-01
④	-0.4000002E-00	0.31929470E-00	-0.9336260E-02	-0.2278235E-01	-0.1183997E-01	-0.1843374E-01
④	-0.3000000E-00	0.4456133E-00	-0.9569179E-02	-0.2591132E-01	-0.1315144E-01	-0.1742968E-01
④	-0.3000003E-00	0.4045888E-00	-0.3646084E-01	-0.2902094E-01	-0.1425692E-01	-0.1643937E-01
④	-0.2500004E-00	0.3623404E-00	-0.2592112E-01	-0.3198322E-01	-0.1628397E-01	-0.1547060E-01
④	-0.2000004E-00	0.3390562E-00	-0.4047294E-01	-0.3442929E-01	-0.1765971E-01	-0.1482946E-01
④	-0.1500005E-00	0.3079450E-00	-0.6172426E-01	-0.3642954E-01	-0.1879517E-01	-0.1352008E-01
④	-0.1000005E-00	0.2794223E-00	-0.9392320E-01	-0.3833718E-01	-0.1961027E-01	-0.1244747E-01
④	-0.5000005E-01	0.8912428E-00	-0.1439871E-00	-0.3916762E-01	-0.2002259E-01	-0.1190417E-01
④	-0.5966444E-00	-0.9999996E-00	-0.2273040E-00	-0.8902446E-01	-0.1995021E-01	-0.1109754E-01
④	0.4299964E-01	0.1122010E-01	-0.2680118E-00	-0.9777930E-01	-0.1930692E-01	-0.1032311E-01
④	-0.2000003E-01	-0.1245222E-01	-0.3405889E-00	-0.9822742E-01	-0.1299648E-01	-0.9520499E-00
④	0.1499990E-00	0.1412535E-00	-0.4871936E-01	-0.3128156E-01	-0.1590668E-01	-0.8850969E-00
④	-0.1999949E-00	-0.1586890E-00	-0.6653512E-01	-0.2542928E-01	-0.1290104E-01	-0.8162778E-00
④	0.2499985E-00	0.1777273E-01	-0.4978080E-01	-0.8089838E-02	-0.7496344E-00	-0.2906384E-00
④	-0.2000004E-00	-0.1909525E-00	-0.1165370E-01	-0.6814931E-02	-0.3420834E-02	-0.6844299E-00
④	0.2499989E-00	0.2233715E-01	-0.1254726E-01	-0.7088340E-02	-0.3528704E-02	-0.6209695E-00
④	-0.3000002E-00	-0.2515104E-01	-0.1651454E-00	-0.7289646E-01	-0.1233667E-01	-0.5509988E-00
④	0.4499983E-00	0.2818372E-01	-0.6933377E-00	-0.4839368E-01	-0.2349723E-01	-0.4962696E-00
④	-0.4999985E-00	-0.3121276E-01	-0.6101037E-00	-0.7037367E-01	-0.3742135E-01	-0.4396663E-00
④	0.5499986E-00	0.3546122E-01	-0.5549397E-01	-0.1117449E-00	-0.5476660E-01	-0.3819197E-00
④	-0.5999988E-00	-0.3981061E-01	-0.5131377E-00	-0.1690282E-00	-0.7634223E-01	-0.3233402E-00
④	0.6499981E-00	0.4466816E-01	-0.4756693E-00	-0.2389607E-00	-0.1032481E-00	-0.2697343E-00
④	-0.6499982E-00	-0.5014545E-01	-0.4423443E-00	-0.3386422E-00	-0.1371180E-00	-0.2147442E-00
④	0.7499985E-01	0.5623939E-01	-0.4152318E-00	-0.4957593E-00	-0.1806769E-00	-0.1597232E-00
④	-0.7999986E-00	-0.6309554E-01	-0.3860010E-00	-0.8151214E-00	-0.2391186E-00	-0.1035349E-00
④	0.8499979E-01	0.7079423E-01	-0.3535231E-01	-0.1274586E-01	-0.3247438E-00	-0.4425835E-01
④	-0.8999984E-00	-0.7943247E-01	-0.3162369E-01	-0.4857039E-01	-0.4728927E-01	-0.2119490E-01
④	0.9499983E-00	0.8912747E-01	-0.2317672E-01	-0.1377799E-01	-0.8667027E-01	-0.9717772E-01
④	-0.9999984E-00	-0.3999946E-01	-0.74731745E-01	-0.1289327E-00	-0.7279714E-01	-0.1826504E-00
④	0.1049997E-01	0.1122012E-02	-0.1071560E-01	-0.3707036E-01	-0.1713285E-01	-0.2792162E-00
④	-0.1049997E-01	-0.1258934E-02	-0.5879129E-01	-0.4881124E-01	-0.1638329E-01	-0.3246309E-00
④	0.1149998E-01	0.1412530E-02	-0.6668890E-01	-0.4334727E-00	-0.6668752E-01	-0.2614587E-00
④	-0.1199998E-01	-0.1584886E-02	-0.3116956E-00	-0.3046422E-00	-0.2903143E-00	-0.1820251E-00
④	0.1249997E-01	0.1778268E-02	-0.5924771E-00	-0.1959879E-00	-0.5451609E-00	-0.1121846E-00
④	-0.1399997E-01	-0.1995254E-02	-0.8874558E-00	-0.1287673E-00	-0.1950674E-00	-0.4697204E-01
④	0.1349997E-01	0.2238708E-02	-0.1095142E-01	-0.7333595E-01	-0.1036302E-01	-0.3975111E-01
④	-0.1399994E-01	-0.2511472E-02	-0.1320436E-01	-0.4446249E-01	-0.1267361E-01	-0.2370349E-01
④	0.1449997E-01	0.2818362E-02	-0.1536391E-01	-0.2717089E-01	-0.1491028E-01	-0.1424543E-01
④	-0.1449997E-01	-0.3162257E-02	-0.1746737E-01	-0.1664566E-01	-0.1709121E-01	-0.8832034E-02
④	0.1549997E-01	0.3546112E-02	-0.1953779E-01	-0.1024825E-01	-0.1922995E-01	-0.5270984E-02
④	-0.1599995E-01	-0.3931049E-02	-0.2116943E-01	-0.6348447E-02	-0.2337314E-01	-0.3246409E-02
④	0.1649997E-01	0.4666803E-02	-0.2362116E-01	-0.3930081E-02	-0.2342081E-01	-0.1997523E-02
④	-0.1699997E-01	-0.5041838E-02	-0.2754642E-01	-0.2450377E-02	-0.2548574E-01	-0.1242404E-02
④	0.1749997E-01	0.5623376E-02	-0.2766489E-01	-0.1523974E-02	-0.2753662E-01	-0.7676166E-03
④	-0.1849997E-02	-0.7079402E-02	-0.3169017E-01	-0.5980714E-03	-0.3160860E-01	-0.3005872E-03
④	0.1949997E-01	0.8912447E-02	-0.3570978E-01	-0.3648073E-02	-0.3363534E-01	-0.1821989E-03
④	-0.1999997E-01	-0.9999934E-02	-0.3771865E-01	-0.1532178E-03	-0.3767746E-01	-0.7910030E-04

		LOG(SENS(REAL(H)))	*****	LOG(SENS4(IM(H)))	*****	LOG(SEN5(AB5(H)))	*****	LOG(SEN6(AB5(H)))	*****	LOG(SEN7(AB5(H)))	*****	LOG(SEN8(AB5(H)))	*****	LOG(SEN9(AB5(H)))	*****	LOG(SEN10(AB5(H)))	*****
		00000000000000000000															
		2.00E 00	1.60E 00	1.20E 00	8.00E -01	4.00E -01	0.00E 00	4.00E -01	8.00E -01	1.20E 00	1.60E 00	2.00E 00					
	-1.0000E 00	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		.	.	.	.	.	.	.	.	.	.	.					
	-7.5000E -01	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		.	.	.	.	.	.	.	.	.	.	.					
	-5.0000E -01	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		.	.	.	.	.	.	.	.	.	.	.					
	-2.5000E -01	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		.	.	.	.	.	.	.	.	.	.	.					
	-5.9605E -07	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		*	*	*	*	*	*	*	*	*	*	*					
	-2.5000E -01	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		.	.	.	.	.	.	.	.	.	.	.					
	5.0000E -01	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		*	*	*	*	*	*	*	*	*	*	*					
	7.5000E -01	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		*	*	*	*	*	*	*	*	*	*	*					
	1.0000E 00	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		*	*	*	*	*	*	*	*	*	*	*					
	1.2500E 00	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		*	*	*	*	*	*	*	*	*	*	*					
	1.5000E 00	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		*	*	*	*	*	*	*	*	*	*	*					
	1.7500E 00	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		*	*	*	*	*	*	*	*	*	*	*					
	2.0000E 00	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....	.....					
		*	*	*	*	*	*	*	*	*	*	*					



## SENSITIVITIES OF ZEROS AND POLES OF TRANSFER FUNCTION

④	ZERO	REAL		IMAG		REAL		SENSITIVITY	
		REAL	IMAG	REAL	IMAG	REAL	IMAG	REAL	IMAG
⑤	1	0.1090900E-29	0.0000000E+00						
⑥	2	-0.0108258E-01	0.0000000E+00						
⑦	3	-0.6876126E+01	0.1258056E+02	0.0000000E+00	0.0000000E+00	0.0000000E+00	0.0000000E+00	0.0000000E+00	0.0000000E+00
⑧	4	-0.6826126E-01	0.1258056E-02	0.0000000E+00	0.0000000E+00	0.0000000E+00	0.0000000E+00	0.0000000E+00	0.0000000E+00

⑨	POLE	REAL		IMAG		REAL		SENSITIVITY	
		REAL	IMAG	REAL	IMAG	REAL	IMAG	REAL	IMAG
⑩	1	-0.5386047E-00	0.0000000E+00	-0.4560962E+01	0.0000000E+00				
⑪	2	-0.2808999E+01	0.1215740E+02	0.3735610E+01	0.1442457E+01				
⑫	3	-0.2808999E+01	0.1215740E+02	-0.3735610E+01	-0.1442457E+01				
⑬	4	-0.7516842E+01	0.0000000E+00	-0.7516842E+01	0.0000000E+00				

X →

C → X → -X → -

X →

$$Y = \mu^2 \cdot k \cdot \sin(\omega t)$$

f = 2, k = 8.8

TGT

NASA PR 0102 HORIZONTAL ATTITUDE CONTROL

MODE  
STEP 10000

V1 1 2 3 1 1 ..

R1 2 3 -1 ..

R2 3 4 1 ..

R3 4 1 1 ..

C1 4 1 F

V2 1 5 6 1 ..

L1 5 1 1 ..

V3 1 6 -1 0 62 111

L2 6 1 1 ..

V4 1 / 1 1 ..

P4 8 7 1 ..

V5 1 8 1 ..

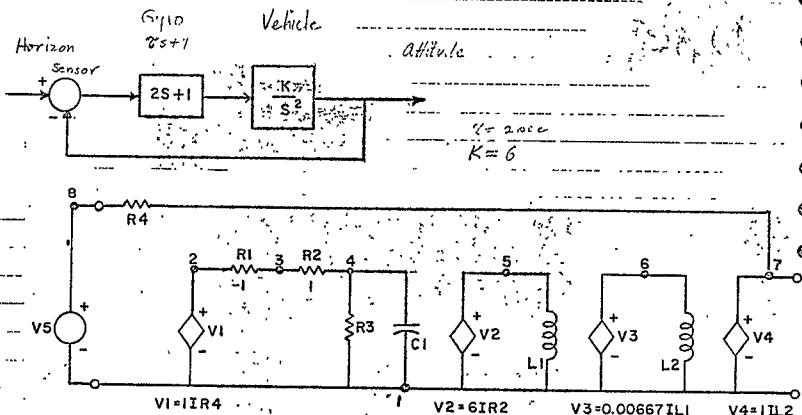
OUTPUT

VV4/VV5/2

FREQ 0. 1 ..

TIME 5 ..

EXECUTE



TRANSFER FNU. T15 : V4/VV5/2

$$\{ \begin{array}{l} 5.00E-01 \quad +1.12E-02 S \quad 0 \\ \end{array} \}$$

$$-H(S) = 0.045 - 0.0001 S$$

$$\{ \begin{array}{l} 4.00E-02 \quad +8.01E-02 S \quad +1.00E-00 S \end{array} \}$$

ZERO OF TRANS. S. 1.0E-01 J1.0E-01

ZERO. REFL. P. 0.112 E-01 J1.0E-01

1 -0.5000E-01 J1.0E-01 J0.0E-00

POLE OF TRANS. FNU C1L1

POLE. IDEAL PAR. 1740. PART

1 -0.4000E-01 J0.1947E-00

2 -0.4000E-01 J0.1947E-00

STEP RESPONSE FUNCTION  
F(T) =

(-0.4002E-01 J 0.1960E 00 ? T  
(-0.5000E\_00 J 0.1021E-00)-E

(-0.4002E+01 J=0.1960E-00)-T  
(-0.5000E 00 J 0.1021E 00 ) E

( 0.0000E 00 J 0.0000E 00 ) T  
(-0.1000E-01-J 0.0000E 00)-E

13

STEP	RESPONSE
TIME	$\Delta V_1/V_{1S}$
-0.0000E-00	1.053161422E-06
0.1000E 01	0.75-2.96195E-01
-0.2000E 01	1.21-9.4611E-01
0.3000E 01	1.36253319E 00
-0.4000E 01	0.53553134E 00
0.5000E 01	1.68763212E 00
-0.6000E 01	0.84644141E 00
0.7000E 01	1.10247437E 01
-0.8000E 01	0.11462402E 01
0.9000E 01	1.27373798E 01
-0.1000E 02	0.40205321E 01
0.1100E 02	0.15095191E 01
-0.1200E 02	0.15212164E 01
0.1300E 02	1.51098599E 01
-0.1400E 02	0.1521195E 01
0.1500E 02	0.15199479E 01
-0.1600E 02	0.15217095E 01
0.1700E 02	0.14717019E 01
-0.1800E 02	0.14132344E 01
0.1900E 02	1.31378844E 01
-0.2000E 02	0.12553779E 01
0.2100E 02	0.111944E 01
-0.2200E 02	0.10393959E 01
0.2300E 02	0.10111159E 01
-0.2400E 02	0.92502228E 00
0.2500E 02	0.85762044E 00
-0.2600E 02	0.80085212E 00
0.2700E 02	0.75508269E 00
-0.2800E 02	0.72451086E 00
0.2900E 02	0.70317845E 00
-0.3000E 02	0.69392162E 00
0.3100E 02	0.70482917E 00
-0.3200E 02	0.72132025E 00
0.3300E 02	0.74701449E 00
-0.3400E 02	0.78136221E 00
0.3500E 02	0.82-9.0404E 00
-0.3600E 02	0.86327349E 00
0.3700E 02	0.89590102E 00
-0.3800E 02	0.95171977E 00
0.3900E 02	0.99419446E 00
-0.4000E 02	1.10233972E 01
0.4100E 02	0.10740356E 01
-0.4200E 02	0.11474626E 01
0.4300E 02	0.11276538E 01
-0.4400E 02	0.11447072E 01
0.4500E 02	0.11556446E 01
-0.4600E 02	0.11284948E 01
0.4700E 02	0.11552443E 01
-0.4800E 02	0.11445846E 01
0.4900E 02	0.11333930E 01
-0.5000E 02	0.11125658E 01

STEP RESPONSE

-4.20E-01 -2.2E-01 -2.1E-02 1.50E-01 3.00E-01 7.00E-01 9.00E-01 1.10E 00 1.30E 00 1.50E 00  
0.0000E 00

5.0000E 00

1.0000E 01

1.5000E 01

2.0000E 01

TOT 2.5000E 01

3.0000E 01

3.5000E 01

4.0000E 01

4.5000E 01

4.9999E 01



SENSITIVITY ANALYSIS

	L0G(FRFQ)	FRFQ	SENS(REF(H))	SENS(IIM(H))	SENS(ABS(H))	SENS(PHI(H))
0	0.0000.00E+00	0.1096766E+01	0.8096766E+00	0.1001704E+01	0.1000853E+01	-0.7794116E-01
1	0.4999995E-01	0.1122018E+01	0.8095220E+00	0.1001352E+01	0.1000677E+01	-0.6974593E-02
2	0.9999996E-01	0.1255925E+01	0.8095778E+00	0.1001746E+01	0.1000537E+01	-0.6298596E-02
3	1.4999997E-01	0.1412352E+01	0.8095432E+00	0.1000853E+01	0.1000426E+01	-0.5578786E-02
4	1.9999998E-01	0.1584892E+01	0.8095166E+00	0.1000677E+01	0.1000339E+01	-0.4986693E-02
5	2.4999999E-01	0.1772785E+01	0.8094950E+00	0.1000534E+01	0.1000269E+01	-0.4486572E-02
6	2.9999997E-01	0.1955261E+01	0.8094779E+00	0.1000427E+01	0.1000214E+01	-0.3981536E-02
7	3.4999996E-01	0.2235719E+01	0.8094636E+00	0.1000340E+01	0.1000170E+01	-0.3556268E-02
8	3.9999995E-01	0.2511183E+01	0.8094522E+00	0.1000265E+01	0.1000134E+01	-0.3175717E-02
9	4.4999994E-01	0.2811340E+01	0.8094464E+00	0.1000214E+01	0.1000107E+01	-0.2835326E-02
10	4.9999993E-01	0.3162274E+01	0.8094374E+00	0.1000170E+01	0.1000085E+01	-0.2530954E-02
11	5.4999992E-01	0.3545113E+01	0.8094313E+00	0.1000134E+01	0.1000067E+01	-0.2258078E-02
12	5.9999991E-01	0.3948107E+01	0.8094273E+00	0.1000107E+01	0.1000053E+01	-0.2015128E-02
13	6.4999990E-01	0.4455429E+01	0.8094236E+00	0.1000084E+01	0.1000042E+01	-0.1798561E-02
14	6.9999989E-01	0.5011165E+01	0.8094212E+00	0.1000066E+01	0.1000032E+01	-0.1602682E-02
15	7.4999988E-01	0.5623425E+01	0.8094192E+00	0.1000053E+01	0.1000027E+01	-0.1431390E-02
16	7.9999987E-01	0.6307593E+01	0.8094174E+00	0.1000042E+01	0.1000021E+01	-0.1276756E-02
17	8.4999986E-01	0.7089445E+01	0.8094158E+00	0.1000032E+01	0.1000016E+01	-0.1138723E-02
18	8.9999985E-01	0.7943267E+01	0.8094153E+00	0.1000027E+01	0.1000013E+01	-0.1015549E-02
19	9.4999984E-01	0.8911491E+01	0.8094145E+00	0.1000021E+01	0.1000010E+01	-0.9056237E-03
20	9.9999983E-01	0.9999978E+01	0.8094139E+00	0.1000016E+01	0.1000009E+01	-0.8075321E-03

	L0G(RF0)	RF0	LOG(SENS(REF(H)))	LOG(SENS(IIM(H)))	LOG(SENS(ABS(H)))	LOG(SENS(PHI(H)))
0	0.0000.00E+00	-0.1068040E+01	-0.9168923E+01	-0.7394078E+03	-0.3692673E+03	-0.2108232E+01
1	0.4999995E-01	0.1122018E+01	0.9171183E+01	-0.5373174E+03	-0.2935510E+03	0.2156479E+01
2	0.9999996E-01	0.1255925E+01	0.9174135E+01	-0.4665251E+03	-0.2313182E+03	0.2204894E+01
3	1.4999997E-01	0.1412352E+01	0.9179559E+01	-0.3702585E+03	-0.1855110E+03	0.2253459E+01
4	1.9999998E-01	0.1584892E+01	0.9177492E+01	-0.2943790E+03	-0.1470074E+03	0.2392016E+01
5	2.4999999E-01	0.1772785E+01	0.9176674E+01	-0.2331162E+03	-0.1159535E+03	0.2350997E+01
6	2.9999997E-01	0.1955261E+01	0.9176190E+01	-0.1835110E+03	-0.9276599E+04	0.2399944E+01
7	3.4999996E-01	0.2138245E+01	0.9176105E+01	-0.1325110E+03	-0.7330286E+04	0.2449001E+01
8	3.9999995E-01	0.2321345E+01	0.9176020E+01	-0.1119539E+03	-0.5480892E+04	0.2498152E+01
9	4.4999994E-01	0.2514390E+01	0.9176031E+01	-0.1474214E+03	-0.3497103E+04	0.2547393E+01
10	4.9999993E-01	0.2717445E+01	0.9176045E+01	-0.1921653E+03	-0.2496454E+04	0.2646095E+01
11	5.4999992E-01	0.2921499E+01	0.9176059E+01	-0.2330919E+03	-0.1421932E+04	0.2695549E+01
12	5.9999991E-01	0.3125553E+01	0.9176073E+01	-0.2749299E+03	-0.1022333E+04	0.2745061E+01
13	6.4999990E-01	0.3330607E+01	0.9176087E+01	-0.3166006E+03	-0.1408113E+04	0.2794624E+01
14	6.9999989E-01	0.3535661E+01	0.9176101E+01	-0.3582735E+03	-0.1772087E+04	0.2844234E+01
15	7.4999988E-01	0.3740715E+01	0.9176115E+01	-0.3998462E+03	-0.2169759E+05	0.2893879E+01
16	7.9999987E-01	0.3945769E+01	0.9176129E+01	-0.4414194E+03	-0.2704920E+05	0.2943563E+01
17	8.4999986E-01	0.4150823E+01	0.9176143E+01	-0.4829525E+03	-0.3299328E+05	0.2993285E+01
18	8.9999985E-01	0.4355876E+01	0.9176157E+01	-0.5244855E+03	-0.3313391E+05	0.3028070E+01

SENSITIVITIES OF ZEROS AND POLES OF TRANSFER FUNCTION

ZERO	REAL	IMAG	REAL	IMAG
1	-0.500000E 00	0.000000E 00	0.000000E 00	0.000000E 00

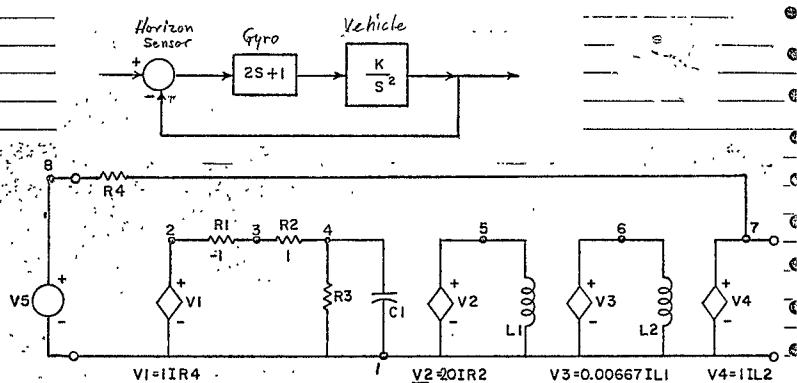
POLE	REAL	IMAG	REAL	IMAG
1	-0.400199E-01	0.196005E 00	-0.400199E-01	0.939173E-01
2	-0.400199E-01	-0.196005E 00	-0.400199E-01	-0.939173E-01

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④  
 NASA PROBLEM RANGER ATTITUDE CONTROL  
 ⑤  
 NONE  
 ⑥ STEP RESPONSE  
 $K = 20$   
 ⑦  
 V1 1 2 1. IR4  
 R1 2 3 -1.  
 R2 3 4 1.  
 R3 4 1 1.  
 C1 4 1 2F  
 V2 1 5 .20 IR2  
 ⑧ L1 5 1 1H  
 V3 6 0.00667 IL1  
 L2 6 1H  
 V4 1 7 1. IL2  
 R4 8 7 1.  
 V5 1 8 1.  
 ⑨ OUTPUT  
 VV4/VV5/V2  
 FREQ 0.0 1.0 0.05  
 ⑩ TIME 50.  
 EXECUTE

---

TRANSFER FUNCTION VV4/VV5/V2



861  
 { 5.00E-01 +1.00E 00 S }  
 H(S) = 2.668E-01 \*-----  
 ( 1.33E-01 +2.67E-01 S +1.00E 00 S )

NUMBER OF LOOPS PER ORDER

1= 5  
2= 3

#### ZERO OF TRANSFER FUNCTION

ZERO REAL PART IMAG. PART

1 -0.50000E 00 0.00000E 00

#### POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

1 -0.13340E 00 0.34001E 00

2 -0.13340E 00 -0.34001E 00

STEP	RESPONSE FUNCTION	STEP	RESPONSE
1	F(T) =		TIME VV4/VV5
2	(-0.1334E 00 J 0.3400E 00 ) T	0.0000E 00	-0.59604445E-07
3	(-0.5000E 00 J-0.1962E 00 ) E	0.1000E 01	0.20948647E 00
4	(-0.1334E 00 J-0.3400E 00 ) T	0.2000E 01	0.59354665E 00
5	(-0.5000E 00 J 0.1962E 00 ) E	0.3000E 01	0.87331635E 00
6	( 0.0000E 00 J 0.0000E 00 ) T	0.4000E 01	0.11023102E 01
7	( 0.1000E 01 J 0.0000E 00 ) E	0.5000E 01	0.12683395E 01
8	( 0.0000E 00 J 0.0000E 00 ) T	0.6000E 01	0.13602829E 01
9	( 0.1000E 01 J 0.0000E 00 ) E	0.7000E 01	0.13908958E 01
10	( 0.0000E 00 J 0.0000E 00 ) T	0.8000E 01	0.13690777E 01
11	( 0.1000E 01 J 0.0000E 00 ) E	0.9000E 01	0.13096333E 01
12	( 0.0000E 00 J 0.0000E 00 ) T	0.1000E 02	0.12282542E 01
13	( 0.1000E 01 J 0.0000E 00 ) E	0.1100E 02	0.11395025E 01
14	( 0.0000E 00 J 0.0000E 00 ) T	0.1200E 02	0.10553808E 01
15	( 0.1000E 01 J 0.0000E 00 ) E	0.1300E 02	0.98454595E 00
16	( 0.0000E 00 J 0.0000E 00 ) T	0.1400E 02	0.93208754E 00
17	( 0.1000E 01 J 0.0000E 00 ) E	0.1500E 02	0.89977700E 00
18	( 0.0000E 00 J 0.0000E 00 ) T	0.1600E 02	0.86663769E 00
19	( 0.1000E 01 J 0.0000E 00 ) E	0.1700E 02	0.88970166E 00
20	( 0.0000E 00 J 0.0000E 00 ) T	0.1800E 02	0.90481949E 00
21	( 0.1000E 01 J 0.0000E 00 ) E	0.1900E 02	0.92741799E 00
22	( 0.0000E 00 J 0.0000E 00 ) T	0.2000E 02	0.95312854E 00
23	( 0.1000E 01 J 0.0000E 00 ) E	0.2100E 02	0.97824574E 00
24	( 0.0000E 00 J 0.0000E 00 ) T	0.2200E 02	0.10000000E 01
25	( 0.1000E 01 J 0.0000E 00 ) E	0.2300E 02	0.10165597E 01
26	( 0.0000E 00 J 0.0000E 00 ) T	0.2400E 02	0.10274897E 01
27	( 0.1000E 01 J 0.0000E 00 ) E	0.2500E 02	0.10325994E 01
28	( 0.0000E 00 J 0.0000E 00 ) T	0.2600E 02	0.10327307E 01
29	( 0.1000E 01 J 0.0000E 00 ) E	0.2700E 02	0.10290546E 01
30	( 0.0000E 00 J 0.0000E 00 ) T	0.2800E 02	0.10228682E 01
31	( 0.1000E 01 J 0.0000E 00 ) E	0.2900E 02	0.10154622E 01
32	( 0.0000E 00 J 0.0000E 00 ) T	0.3000E 02	0.10080328E 01
33	( 0.1000E 01 J 0.0000E 00 ) E	0.3100E 02	0.10013970E 01
34	( 0.0000E 00 J 0.0000E 00 ) T	0.3200E 02	0.99612618E 00
35	( 0.1000E 01 J 0.0000E 00 ) E	0.3300E 02	0.99258500E 00
36	( 0.0000E 00 J 0.0000E 00 ) T	0.3400E 02	0.99270796E 00
37	( 0.1000E 01 J 0.0000E 00 ) E	0.3500E 02	0.99034858E 00
38	( 0.0000E 00 J 0.0000E 00 ) T	0.3600E 02	0.99118954E 00
39	( 0.1000E 01 J 0.0000E 00 ) E	0.3700E 02	0.99285364E 00
40	( 0.0000E 00 J 0.0000E 00 ) T	0.3800E 02	0.99492542E 00
41	( 0.1000E 01 J 0.0000E 00 ) E	0.3900E 02	0.99714911E 00
42	( 0.0000E 00 J 0.0000E 00 ) T	0.4000E 02	0.99915910E 00
43	( 0.1000E 01 J 0.0000E 00 ) E	0.4100E 02	0.10007954E 01
44	( 0.0000E 00 J 0.0000E 00 ) T	0.4200E 02	0.10015960E 01
45	( 0.1000E 01 J 0.0000E 00 ) E	0.4300E 02	0.10026188E 01
46	( 0.0000E 00 J 0.0000E 00 ) T	0.4400E 02	0.10028229E 01
47	( 0.1000E 01 J 0.0000E 00 ) E	0.4500E 02	0.10026522E 01
48	( 0.0000E 00 J 0.0000E 00 ) T	0.4600E 02	0.10022144E 01
49	( 0.1000E 01 J 0.0000E 00 ) E	0.4700E 02	0.10016222E 01
50	( 0.0000E 00 J 0.0000E 00 ) T	0.4800E 02	0.10009813E 01
51	( 0.1000E 01 J 0.0000E 00 ) E	0.4900E 02	0.10003757E 01
52	( 0.0000E 00 J 0.0000E 00 ) T	0.5000E 02	0.99986970E 00

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**STEP RESPONSE**

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-6.00E-01 -4.00E-01 -2.00E-01 0.00E 00 2.00E-01 4.00E-01 6.00E-01 8.00E-01 1.00E 00 1.20E 00 1.40E 00

草  
吉  
草  
吉

5-0000E 00

1.0000E\_01

<sup>6</sup> See also the discussion of the relationship between the two concepts in the section on "The Concept of Social Capital."

1.5000E 01

2.5000E 01

<sup>22</sup> See also the discussion of the relationship between the two in the section on the "Economic Crisis."

3.0000E 01 \* . . . . .

3 3.5000E 01

家水

4.5000E 01

• . • . • . • . • . • . • . • . • .

4.9999E 01 \* .....

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			LOG(FREQ)	FREQ	20.*LOG(ABS(H))	PHI(H)	ABS(H)	LOG(ABS(H))
②			0.000000E_00	0.1000000E_01	-0.2739093E_02	-0.9211026E_02	0.4270237E_01	-0.1369557E_01
③			0.4999995E-01	0.1122018E_01	-0.2840099E_02	-0.918379E_02	0.3801452E_01	-0.1420050E_01
④			0.5999990E-01	0.1253235E_01	-0.2940898E_02	-0.9168105E_02	0.3384930E_01	-0.1470449E_01
⑤			0.1499979E_00	0.1412337E_01	-0.3041531E_02	-0.9149976E_02	0.3014620E_01	-0.1520766E_01
⑥			0.1999988E_00	0.1584928E_01	-0.3142038E_02	-0.913772E_02	0.2685222E_01	-0.1571019E_01
⑦			0.2499979E_00	0.1778278E_01	-0.3242439E_02	-0.9119301E_02	0.2392103E_01	-0.1621220E_01
⑧			0.2999979E_00	0.1995761E_01	-0.3342738E_02	-0.9109381E_02	0.2131911E_01	-0.1671379E_01
⑨			0.3499979E_00	0.2212319E_01	-0.3442939E_02	-0.9099561E_02	0.1884888E_01	-0.1721560E_01
⑩			0.4499968E_00	0.2521321E_01	-0.3543213E_02	-0.9084564E_02	0.1691972E_01	-0.1771664E_01
⑪			0.4999995E_00	0.2813208E_01	-0.3643369E_02	-0.9075396E_02	0.1507649E_01	-0.1821495E_01
⑫			0.5499995E_00	0.3162274E_01	-0.3743497E_02	-0.9067201E_02	0.1343549E_01	-0.1871749E_01
⑬			0.5999994E_00	0.3540130E_01	-0.3843597E_02	-0.9059903E_02	0.1197292E_01	-0.1921799E_01
⑭			0.6499994E_00	0.3981047E_01	-0.3943677E_02	-0.9053398E_02	0.1066990E_01	-0.1971839E_01
⑮			0.6999979E_00	0.45669129E_01	-0.4043741E_02	-0.9047597E_02	0.9508595E_02	-0.2021871E_01
⑯			0.5011656E_01	0.4114793E_02	-0.9042422E_02	-0.8474287E_02	-0.2071897E_01	
⑰			0.7499993E_00	0.5623405E_01	-0.42424831E_02	-0.9037611E_02	0.7552367E_02	-0.2121916E_01
⑱			0.7999992E_00	0.6302963E_01	-0.43433864E_02	-0.9033701E_02	0.6730810E_02	-0.2171932E_01
⑲			0.8499992E_00	0.70774945E_01	-0.44443889E_02	-0.9030043E_02	0.5998664E_02	-0.2221945E_01
⑳			0.9499991E_00	0.79493267E_01	-0.45433909E_02	-0.9026779E_02	0.5346190E_02	-0.2271556E_01
㉑			0.9999991E_00	0.8912491E_01	-0.46463925E_02	-0.9023869E_02	0.4765710E_02	-0.2321963E_01
㉒			0.9999990E_00	0.9999978E_01	-0.4743938E_02	-0.9021272E_02	0.4246488E_02	-0.2371969E_01

			LOG(FREQ)	FREQ	20.*LOG(ABS(H))
㉓			-5.22E 01	-4.97E 01	-4.72E 01
㉔			-4.47E 01	-4.22E 01	-3.97E 01
㉕			-3.72E 01	-3.47E 01	-3.22E 01
㉖			-2.97E 01	-2.72E 01	
㉗			0.0000E 00		*
㉘			2.5000E-01		*
㉙			5.0000E-01		*
㉚			7.5000E-01		*
㉛			1.0000E 00		*

			LOG(FREQ)	FREQ	PHI
㉜			-2.00E 02	-1.60E 02	-1.20E 02
㉝			-8.00E 01	-4.00E 01	0.00E 00
㉞			4.00E 01	8.00E 01	1.20E 02
㉟			1.60E 02	2.00E 02	
㉟			0.0000E 00		*
㉟			2.5000E-01		*
㉟			5.0000E-01		*
㉟			7.5000E-01		*
㉟			1.0000E 00		*

## SENSITIVITY ANALYSIS \*\*

	LOG(FREQ)	FREQ	ABS(SENS(H))	PHI(SENS(H))	L08(ABS(SENS(H)))
④	0.000000E+00	0.100000E+01	0.1002481E+01	0.2539496E+01	0.100000E+02
⑤	0.4999995E-01	0.1122018E+01	0.1001969E+01	0.2173139E+01	0.1544312E+03
⑥	0.9999990E-01	0.1254945E+01	0.1001563E+01	0.1949593E+01	0.1078304E+03
⑦	0.1499995E+00	0.1412177E+01	0.1001224E+01	0.1724777E+01	0.1499217E+03
⑧	0.1999998E+00	0.1594892E+01	0.1000986E+01	0.15150767E+01	0.4480463E+03
⑨	0.2499998E+00	0.1777278E+01	0.1000783E+01	0.1369333E+01	0.3399048E+03
⑩	0.2999997E+00	0.1995261E+01	0.1000621E+01	0.1220200E+01	0.2695443E+03
⑪	0.3499997E+00	0.2232179E+01	0.1000493E+01	0.1087374E+01	0.2140759E+03
⑫	0.3999996E+00	0.2515179E+01	0.1000359E+01	0.9680744E+00	0.1876778E+03
⑬	0.4499995E+00	0.2814230E+01	0.1000235E+01	0.8523505E+00	0.1584122E+03
⑭	0.4999995E+00	0.3162247E+01	0.1000120E+01	0.7495715E+00	0.1072591E+03
⑮	0.5499995E+00	0.3548130E+01	0.1000196E+01	0.6593408E+00	0.823117E-04
⑯	0.5999994E+00	0.3981607E+01	0.1000152E+01	0.6112304E+00	0.6769123E-04
⑰	0.6499994E+00	0.4466682E+01	0.1000124E+01	0.5447939E+00	0.5583947E-04
⑱	0.6999993E+00	0.5011610E+01	0.1000097E+01	0.4855469E+00	0.4224383E-04
⑲	0.7499993E+00	0.5623340E+01	0.1000077E+01	0.4326795E+00	0.3354691E-04
⑳	0.7999992E+00	0.6309693E+01	0.1000061E+01	0.3856919E+00	0.2650642E-04
㉑	0.8499992E+00	0.7079445E+01	0.1000049E+01	0.3450785E+00	0.2112243E-04
㉒	0.8999991E+00	0.7913735E+01	0.1000038E+01	0.3063000E+00	0.1556670E-04
㉓	0.9499991E+00	0.8824241E+01	0.1000030E+01	0.2223500E+00	0.1289241E-04
㉔	0.9999990E+00	0.9813731E+01	0.1000023E+01	0.1505032E+00	0.1000000E+00

	LOG(FREQ)	PHL(SENS)	40 <sup>0</sup>
①	-2.00E 02 -1.60E 02 -1.20E 02 -8.00E 01 -4.00E 01 0.00E 00	4.00E 01 8.00E 01 1.20E 02 1.60E 02 2.00E 02	
②	0.0000E 00	.	*
③	.	*	.
④	.	*	.
⑤	.	*	.
⑥	2.5000E-01	.....	.....
⑦	.	*	.
⑧	.	*	.
⑨	.	*	.
⑩	5.0000E-01	.....	.....
⑪	.	*	.
⑫	.	*	.
⑬	.	*	.
⑭	7.5000E-01	.....	.....
⑮	.	*	.
⑯	.	*	.
⑰	.	*	.
⑲	1.0000E 00	.....	.....

④ - SENSITIVITIES OF ZEROS AND POLES OF TRANSFER FUNCTION

④	ZERO	REAL	IMAG	REAL	SENSITIVITY
④					IMAG
④	1	-0.4999998E 00	0.0000000E 00	-0.4468126E-06	0.0000000E 00

④	POLE	REAL	IMAG	REAL	SENSITIVITY
④					IMAG
④	1	-0.1333996E 00	0.3400062E 00	-0.1333997E 00	0.1438336E 00
④	2	-0.1333996E 00	-0.3400062E 00	-0.1333997E 00	-0.1438336E 00

④ 3

④ NASAP PROBLEM RANGER ATTITUDE CONTROL

④ NONE  
④ STEP RESPONSE

④ V1 1 2 1. IR4  
④ R1 2 3 -1.  
④ R2 3 4 1.  
④ R3 4 1 1.  
④ C1 4 -1 -2F  
④ V2 1 5 75 IR2  
④ L1 5 1 1H  
④ V3 1 6 0.00667 IL1  
④ L2 6 1 1H  
④ V4 1 7 1. IL2  
④ R4 -8 7 1.  
④ V5 1 8 1.  
④ OUTPUT  
④ VV4/VV5/V2  
④ FREQ 0.0 1.0 0.0 0.05  
④ TIME 50.  
④ EXECUTE

K=75

④ TRANSFER-FUNCTION—VV4/VV5/V2 NUMBER-OF-LOOPS-PER-ORDER

④ 1= 5  
④ 2= 3

④ { 5.00E-01 +1.00E 00 S }

④ H(S)= 1.000E 00\*

④ { 5.00E-01 +1.00E 00 S +1.00E 00 S }

④ ZERO OF TRANSFER FUNCTION

④ ZERO REAL PART IMAG. PART

④ 1 -0.50000E 00 0.00000E 00

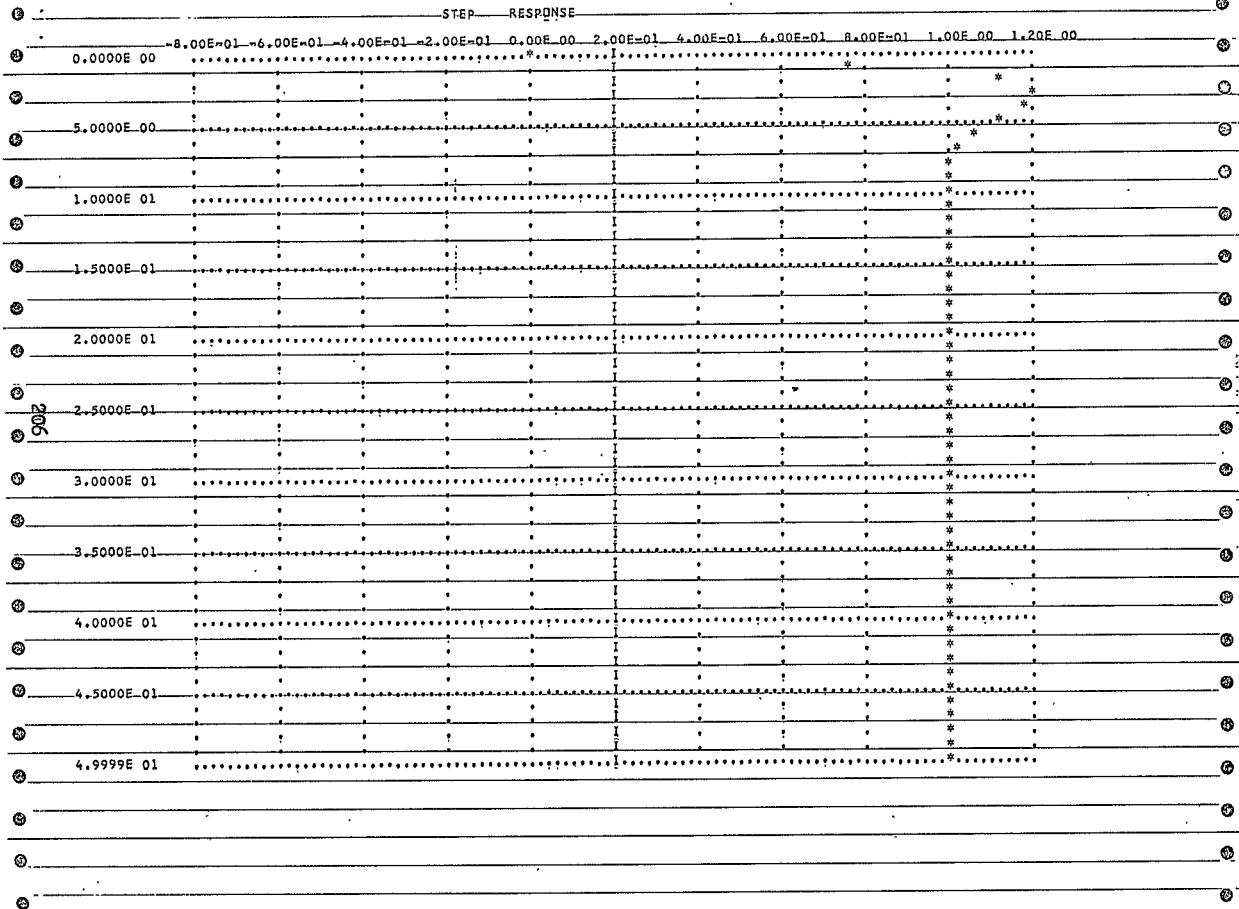
④ POLE OF TRANSFER FUNCTION

④ POLE REAL PART IMAG. PART

④ 1 -0.50025E 00 0.50000E 00

④ 2 -0.50025E 00 -0.50000E 00

STEP	RESPONSE-FUNCTION	TIME	STEP	RESPONSE
①	F(T)			VV4/VV5
②	(-0.5002E-00 J-0.5002E-00) T	0.0000E 00		-0.2384185BE-06
③	(-0.5000E 00 J-0.5000E 00) E	0.1000E 01		0.75870979E-00
④	(-0.5002E 00 J-0.5000E 00) T	0.2000E 01		0.11108913E 01
⑤	(-0.5000E-00 J-0.5002E-00) E	0.3000E 01		0.12067432E 01
⑥	(-0.5000E-00 J-0.5002E-00) T	0.4000E 01		0.11792612E 01
⑦	(-0.5000E 00 J-0.5000E 00) E	0.5000E 01		0.11147690E 01
⑧	(-0.5000E 00 J-0.5000E 00) T	0.6000E 01		0.10562334E 01
⑨	(-0.5000E 00 J-0.5000E 00) E	0.7000E 01		0.10176487E-01
⑩	( 0.1000E 01 J 0.0000E 00 ) E	0.8000E 01		0.99810702E 00
⑪		0.9000E 01		0.99149549E 00
⑫		0.1000E 02		0.99164456E 00
⑬		0.1100E 02		0.99423426E 00
⑭		0.1200E 02		0.99693567E 00
⑮		0.1300E 02		0.99885851E 00
⑯		0.1400E 02		0.99991173E 00
⑰		0.1500E 02		0.10003232E 01
⑱		0.1600E 02		0.10003786E 01
⑲		0.1700E 02		0.10002832E 01
⑳		0.1800E 02		0.10001621E 01
㉑		0.1900E 02		0.10000677E 01
㉒		0.2000E 02		0.10000124E 01
㉓		0.2100E 02		0.99998844E 00
㉔		0.2200E 02		0.99998283E 00
㉕		0.2300E 02		0.99998581E 00
㉖		0.2400E 02		0.99999106E 00
㉗		0.2500E 02		0.99999553E 00
㉘		0.2600E 02		0.99999839E 00
㉙		0.2700E 02		0.99999976E 00
㉚		0.2800E 02		0.10000000E 01
㉛		0.2900E 02		0.10000000E 01
㉜		0.3000E 02		0.99999940E 00
㉝		0.3100E 02		0.99999920E 00
㉞		0.3200E 02		0.99999958E 00
㉟		0.3300E 02		0.99999952E 00
㉟		0.3400E 02		0.99999946E 00
㉟		0.3500E 02		0.99999946E 00
㉟		0.3600E 02		0.99999946E 00
㉟		0.3700E 02		0.99999946E 00
㉟		0.3800E 02		0.99999946E 00
㉟		0.3900E 02		0.99999952E 00
㉟		0.4000E 02		0.99999952E 00
㉟		0.4100E 02		0.99999952E 00
㉟		0.4200E 02		0.99999952E 00
㉟		0.4300E 02		0.99999952E 00
㉟		0.4400E 02		0.99999952E 00
㉟		0.4500E 02		0.99999952E 00
㉟		0.4600E 02		0.99999952E 00
㉟		0.4700E 02		0.99999952E 00
㉟		0.4800E 02		0.99999952E 00
㉟		0.4900E 02		0.99999952E 00
㉟		0.5000E 02		0.99999952E 00



LLOC(FREQ)	FREQ	20+LOG(ABS(X))	PHI(H)	ABS(H)	LNU ABSI(H)
0.4750000E-00	0.1000000E-01	-0.1593262E-02	-0.8538020E-02	0.1597236E-01	-0.7696300E-00
0.4999995E-00	0.1122010E-01	-0.1693796E-02	-0.8599842E-02	0.1422662E-01	-0.8469891E-00
0.4999990E-01	0.1258925E-01	-0.1794226E-02	-0.8635066E-02	0.1467319E-01	-0.8469861E-00
0.4999995E-01	0.1412337E-01	-0.1894717E-02	-0.8675181E-02	0.1467319E-01	-0.8469861E-00
0.4999990E-01	0.1564892E-01	-0.1994847E-02	-0.8715826E-02	0.1467319E-01	-0.8469861E-00
0.1999998E-00	0.1778278E-01	-0.2094978E-02	-0.8746205E-02	0.1467319E-01	-0.8469861E-00
0.2499998E-00	0.1995261E-01	-0.2195110E-02	-0.8776545E-02	0.1467319E-01	-0.8469861E-00
0.4999977E-00	0.2295382E-01	-0.2295243E-02	-0.8795700E-02	0.1467319E-01	-0.8469861E-00
0.3499997E-00	0.2295446E-01	-0.2395446E-02	-0.8817799E-02	0.1467319E-01	-0.8469861E-00
0.3999997E-00	0.2395480E-01	-0.2495480E-02	-0.8837846E-02	0.1467319E-01	-0.8469861E-00
0.4499995E-00	0.2818380E-01	-0.2495515E-02	-0.8855151E-02	0.1467319E-01	-0.8469861E-00
0.4999995E-00	0.3162274E-01	-0.2595561E-02	-0.8875515E-02	0.1467319E-01	-0.8469861E-00
0.5499995E-00	0.3548130E-01	-0.2695705E-02	-0.8891260E-02	0.1488995E-01	-0.1347854E-01
0.5999994E-00	0.3981067E-01	-0.2795750E-02	-0.8905277E-02	0.1400582E-01	-0.1397876E-01
0.6499994E-00	0.4466829E-01	-0.2895787E-02	-0.8987774E-02	0.13665237E-01	-0.1477894E-01
0.6999993E-00	0.5011855E-01	-0.2995816E-02	-0.8908896E-02	0.1317731E-01	-0.1547910E-01
0.7499993E-00	0.56223405E-01	-0.3095837E-02	-0.8921105E-02	0.1291313E-01	-0.1547910E-01
0.7999992E-00	0.6309563E-01	-0.3195856E-02	-0.8933464E-02	0.1262389E-01	-0.1547910E-01
0.8499992E-00	0.7079445E-01	-0.3295875E-02	-0.8945823E-02	0.1233437E-01	-0.1547910E-01
0.8999991E-00	0.7945920E-01	-0.3395894E-02	-0.8949252E-02	0.1204472E-01	-0.1547910E-01
0.9499991E-00	0.8849149E-01	-0.3495918E-02	-0.8954872E-02	0.11786710E-01	-0.1547910E-01
0.9999990E-00	0.9828671E-01	-0.3595892E-02	-0.89594355E-02	0.11592393E-01	-0.1547910E-01

$\text{LOG(FREQ)} = \frac{20, * \text{LOG}(\text{ABS}(H))}{-4,07E\ 01 - 3,82E\ 01 - 3,57E\ 01 - 3,32E\ 01 - 3,07E\ 01 - 2,82E\ 01 - 2,57E\ 01 - 2,33E\ 01 - 2,08E\ 01 - 1,83E\ 01 - 1,58E\ 01}$

LOC(FREQ) -2.00E 02 -1.60E 02 -1.20E 02 -8.00E 01 -4.00E 01 0.00E 00 PHI 4.00E 01 8.00E 01 -1.20E 02 1.60E 02 2.00E 02

## SENSITIVITY-ANALYSIS

	LOG(FREQ)	FREQ	ABS(SENS(H))	PHI(SENS(H))	LOG(ABS(SENS(H)))
+	-0.000000E 00	-0.100000E 01	0.9999137E 00	-0.9161669E-01	-0.3748494E-04
+	0.4999995E-01	0.1122018E 01	0.9999446E 00	0.8158376E 01	-0.2412638E-04
+	0.9999990E-01	0.1258925E 01	0.9999644E 00	-0.2266209E 01	-0.2266209E-04
+	0.1499999E 00	0.1412537E 01	0.9999767E 00	0.6742727E 01	-0.1012135E-04
+	0.1999998E 00	0.1584892E 01	0.9999824E 00	-0.5740115E 01	-0.466272445E-05
+	0.2499998E 00	0.1778271E 01	0.9999881E 00	0.1327315E 01	-0.3448864E-05
+	0.2999995E 00	0.1973515E 01	0.9999932E 00	-0.4572395E 01	-0.2892937E-05
+	0.3499995E 00	0.2228719E 01	0.9999986E 00	0.4078719E 01	-0.1896797E-05
+	0.3999996E 00	0.2511885E 01	0.9999970E 00	-0.3634531E 01	-0.1320186E-05
+	0.4499996E 00	0.2818390E 01	0.9999980E 00	0.3238844E 01	-0.8801237E-06
+	0.4999995E 00	0.3162274E 01	0.9999986E 00	-0.2886308E 01	-0.595377E-06
+	0.5499995E 00	0.3548130E 01	0.9999989E 00	0.2572201E 01	-0.4659474E-06
+	0.5999994E 00	0.3981067E 01	0.9999990E 00	-0.2292319E 01	-0.4100816E-06
+	0.6499994E 00	0.4466829E 01	0.1000000E 01	0.2046290E 01	-0.3600000E 00
+	0.6999993E 00	0.5011856E 01	0.1000000E 01	-0.1626225E 01	-0.2888597E-07
+	0.7499993E 00	0.5623405E 01	0.9999990E 00	0.14465123E 01	-0.2288597E-07
+	0.7999992E 00	0.6335054E 01	0.9999990E 00	-0.1288829E 01	-0.1035439E-06
+	0.8499992E 00	0.7079445E 01	0.9999998E 00	0.1023722E 01	-0.0000000E 00
+	0.8999991E 00	0.7842675E 01	0.1000000E 01	-0.9123837E 00	-0.517193E-07
+	0.9499991E 00	0.8912491E 01	0.1000000E 01	0.1023722E 01	-0.0000000E 00
+	0.9999990E 00	0.9999978E 01	0.9999999E 00	-0.9123837E 00	-0.517193E-07

$\text{LOG}(\text{FPTQ})$        $\text{LOG}(\text{ABS}(\text{SENS}))$

LOG(EREQ) .PHI(SENS)  
-2.00E 02 -1.60E 02 -1.20E 02 -8.00E 01 -4.00E 01 0.00E 00 4.00E 01 8.00E 01 1.20E 02 1.60E 02 2.00E 02

④ SENSITIVITIES OF ZEROES AND POLES OF TRANSFER FUNCTION

ZERO	REAL	IMAG	SENSITIVITY	
	REAL	IMAG	REAL	IMAG
1	-0.4909999E-00	0.0000000E-00	-0.1191500E-06	0.0000000E-00

POLE	REAL	IMAG	SENSITIVITY	
	REAL	IMAG	REAL	IMAG
1	-0.5002484E-00	0.5000001E-00	-0.5002484E-00	-0.2489090E-03
2	-0.5002484E-00	-0.5000001E-00	-0.5002484E-00	0.2489090E-03

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CHAPTER VII  
SPECIAL CONTROL SYSTEM EXAMPLES

In the previous chapters of this manual we have used a variety of control circuit problems to illustrate various facets of computer aided design with NASAP. The examples were basic to control theory with explicit or implicit connection to aerospace applications. In this final chapter we have selected a few control system examples that emphasize the capabilities and also the limitations of the present version of NASAP as modified at the Moore School. From these examples it will be apparent that, in general, NASAP can be an effective aid to the control system engineer dealing with moderately complex linear control systems.

## VIIA EXAMPLE INVOLVING TIME DOMAIN APPROXIMATION

Approximation of system response is a long standing problem related to control system analysis and synthesis. It is often necessary to approximate a system before suitable compensation can be applied. Eisenberg [EI 1] has described a technique for approximately identifying high order system responses based on certain characteristic responses to a unit step input. Many feedback systems exhibit time domain response to a unit step input that includes an overshoot followed by variations which subsequently settle to a steady state value. The approximate system response is generated by a closed loop transfer function whose open loop transfer function contains a transport lag and a first order lag. This technique only requires knowledge of the first peak overshoot, the time to peak and the settling time of the unknown system response.

We represent the unknown response by that of a unity-feedback control system with  $G(s)$  representing the plant. The general form of  $G(s)$  is assumed to be

$$G(s) = \frac{K_p e^{-sT_d}}{\tau s + 1} \quad (7.1)$$

This particular approximation is quite convenient and yields the system transfer function

$$\frac{C(s)}{R} = \frac{G(s)}{1 + G(s)} = \frac{\frac{K_p}{\tau} e^{-sT_d}}{s + \frac{1}{\tau} + \frac{K_p}{\tau} e^{-sT_d}} \quad (7.2)$$

To facilitate further consideration of this expression, we normalize it with respect to  $T_d$ . Defining

$$\bar{s} = T_d s = \sigma T_d + j\omega T_d = \bar{\sigma} + j\bar{\omega}$$

we can write

$$\frac{C(s)}{R(s)} = \frac{\alpha e^{-\bar{s}}}{\bar{s} + \beta + \alpha e^{-\bar{s}}} \quad (7.3)$$

where

$$\alpha = \frac{K_p T_d}{\tau}, \quad \beta = \frac{T_d}{\tau}, \quad \alpha = K_p \beta$$

The problem of system approximation is now reduced to that of determining the parameters  $\alpha$  and  $\beta$  such that the denominator of (7.3) has a pair of clearly dominant complex conjugate roots which will exhibit a second-order response. In particular it is desired that the responses shown in Fig. 7.1 be similar, for the set of conditions

$$M_h \approx M_s = M, \quad T_{ph} \approx T_{ps} = T, \quad T_{sh} \approx T_{ss} = T_s \quad (7.4)$$

where the values of  $M_h$ ,  $T_{ph}$ , and  $T_{sh}$  are measured from the unknown system step response.

One way of accomplishing the selection of  $\alpha$  and  $\beta$  is to use the generalized curves found in [EI 1]. An alternative is to find a suitable electric circuit model for  $G(s)$  in (7.1) and then use NASAP. The new feature here is the exponential,  $e^{-sT_d}$ . We can obtain a rational function representation for this exponential by a Padé approximation. Specifically we use the biquadratic

$$\frac{s^2 - as + b}{s^2 + as + b}$$

In Chapter III, we showed how to obtain a ladder network for this with negative as well as positive elements. The input impedance of the network is the desired circuit model.

To illustrate this approximation technique we again consider the seventh order system with an open loop transfer function

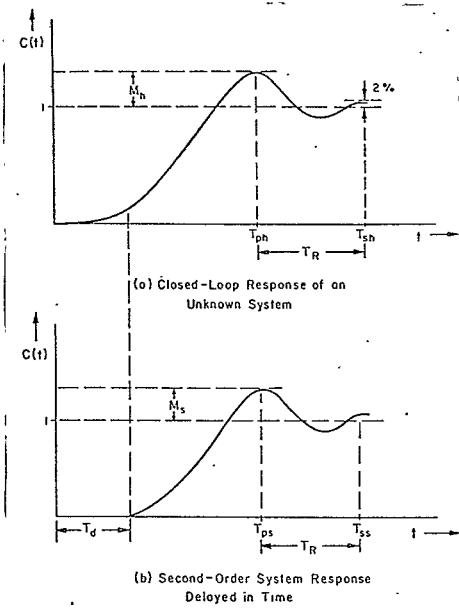


Fig. 7.1 System Approximation by Time Domain Response  
(Subscripts "h" and "s" refer to higher order  
and second order system respectively)

$$G(s) = \frac{1}{(6s + 1)(2s + 1)^3(s + 1)^3} \quad (7.5)$$

The step response for this in a unity feedback system was given in Chapter V as the "Eisenberg" problem. Consideration of the actual closed-loop, unit step response of this system, the lower response curve shown in Fig. 7.2 from [EI 1], yields the three data point  $M_h = 31.7\%$ ,  $T_{ph} = 20s$ , and  $T_{sh} = 55.4s^2$ .

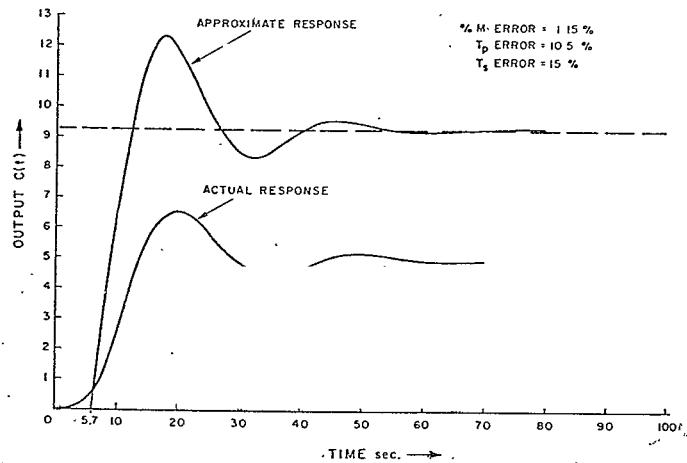


Fig. 7.2 Comparison of System Time Responses.

To obtain the approximate system parameters we must fall back on the well known relationships for a second-order system

$$M_s = \frac{(\frac{\pi}{\zeta\omega_n})}{\sqrt{1 - \zeta^2}} \quad (7.6)$$

$$T_{ps} = \frac{\pi}{\omega_n \sqrt{1 - \zeta^2}} \quad (7.7)$$

$$T_{ss} = \frac{4}{\zeta\omega_n} \quad (\text{for a } 2\% \text{ deviation of the response envelope from the steady-state value}) \quad (7.8)$$

The determination of  $\zeta$  and  $\omega_n$  and ultimately location of the dominant roots involves use of (7.4). Since  $M_h \approx M_s$ , where  $M_h$  is a measured quantity, we have the numerical value  $M_s$ . Hence the value of  $\zeta$  can either be computed from (7.6) as

$$\zeta = \frac{|\ln M|}{\sqrt{(\ln M)^2 + \tau^2}} \quad (7.9)$$

or it can be obtained from a universal curve of  $M$  versus  $\xi$ . Next to determine  $\omega_n$ . Note from Fig. 7.1 a and b and (7.4) that

$$T_R = T_{sh} - T_{ph} = T_{ss} - T_{ps} \quad (7.10)$$

Substituting the required values and solving for  $\omega_n$  gives

$$\omega_n = \frac{4}{T_R \zeta} + \frac{\tau}{T_R \sqrt{1 - \zeta^2}} \quad (7.11)$$

Since  $\zeta$  is known  $\omega_n$  can be expressed in terms of measured quantities as

$$\omega_n = \frac{\sqrt{(\ln M)^2 + \tau^2}}{T_R} \left( \frac{4}{|\ln M|} - 1 \right) \quad (7.12)$$

Next we find  $T_d$  as a function of the measured parameters. From Fig. 7.1a and b

$$T_d = T_{ph} - T_{ps} \quad (7.13)$$

where  $T_{ph}$  is a measured quantity and  $T_{ps}$  can be obtained from (7.7). Substituting (7.7) into (7.13) gives

$$T_d = \frac{\frac{4T_{ph}}{4} - |\ln M| T_{sh}}{\frac{4}{4} - |\ln M|} \quad (7.14)$$

so that the values of  $\zeta$ ,  $\omega_n$ , and  $T_d$  are completely specified from the key characteristics of the unknown system response. Thus the values of the approximate system parameters are found to be  $T_d = 5.7\text{s}$ ,  $\tau = 89.7\text{s}$ , and

$K_p = 13.1$ . These give an approximate open-loop transfer function of

$$G(s) = \frac{13.1e^{-5.7s}}{89.7s + 1} \quad (7.15)$$

The closed-loop response corresponding to this  $G(s)$  obtained in the reference is shown in the upper response curve in Fig. 7.2. The percentage differences between the indicated characteristics of the two responses are also shown. It should be pointed out that although the responses match quite closely with respect to  $M$ ,  $T_p$ , and  $T_s$  the steady-state levels will not match. This mismatch is unimportant because the steady-state levels can always be matched exactly by simply adding gain outside the feedback loop. Also the gain of the unknown system response is easily computed from the measurable steady-state level, ( $c_{ss}(t)$ ), via the relationship

$$K = \frac{c_{ss}(t)}{1 - \bar{c}_{ss}(t)} \quad (7.16)$$

The NASAP print out and step response for (7.15) is shown in Figs. 7.3 and 7.4.

It should be noted for aerospace control applications a distinct advantage of the technique is that it is not necessary to open any feedback loops to effect the approximation. Thus, the system response can be approximated without interrupting normal operation. This is in contrast to some methods of system approximation, this technique is not hampered by the rare cases where the open-loop response is oscillatory. There are other methods available for system approximation that can be used on a closed-loop basis but these methods usually require that the system gain be increased until critical cycling is attained (oscillations). For many practical applications, however, it is not advisable to bring a control system to the verge of instability.

STEP	REFERENCE FUNCTION	STEP	REFERENCE
F(T) =		TIME	VVB/VV1
		0.0000E 00	0.23341956E-06
①	(-0.8220E-01 J 0.2209E 00 ) T	0.1000E 01	0.19248191E-01
②	(-0.1513E 00 J 0.3452E 00 ) E	0.2000E 01	-0.70372520E-07
③	(-0.8220E-01 J-0.2209E 00 ) T	0.3000E 01	-0.27367164E-01
④	(-0.1513E 00 J-0.3452E 00 ) E	0.4000E 01	-0.23492847E-01
⑤	(-0.1045E 01 J 0.0000E 00 ) T	0.5000E 01	-0.29194755E-03
⑥	(-0.1973E 00 J 0.1710E-07 ) E	0.6000E 01	0.45559823E-01
⑦	( 0.0000E 00 J 0.0000E 00 ) T	0.7000E 01	0.10725962E 00
⑧	( 0.4999E 00 J-0.1577E-07 ) E	0.8000E 01	0.17949171E 00
⑨	Steady state Error C <sub>ss</sub> = 0.500	0.9000E 01	0.25706337E 00
⑩		0.1000E 02	0.33923371E 00
⑪		0.1100E 02	0.40949364E 00
⑫		0.1200E 02	0.47790761E 00
⑬		0.1300E 02	0.53568549E 00
⑭		0.1400E 02	0.58467632E 00
⑮		0.1500E 02	0.62106609E 00
⑯		0.1600E 02	0.66375956E 00
⑰		0.1700E 02	0.69577076E 00
⑱		0.1800E 02	0.66266614E 00
⑲		0.1900E 02	0.65713841E 00
⑳		0.2000E 02	0.64448052E 00
㉑		0.2100E 02	0.62036781E 00
㉒		0.2200E 02	0.60015207E 00
㉓		0.2300E 02	0.58070273E 00
㉔		0.2400E 02	0.55634976E 00
㉕		0.2500E 02	0.53280711E 00
㉖		0.2600E 02	0.51114539E 00
㉗		0.2700E 02	0.49199191E 00
㉘		0.2800E 02	0.47651201E 00
㉙		0.2900E 02	0.46339201E 00
㉚		0.3000E 02	0.45541633E 00
㉛		0.3100E 02	0.45096495E 00
㉜		0.3200E 02	0.44252499E 00
㉝		0.3300E 02	0.43083511E 00
㉞		0.3400E 02	0.42586171E 00
㉟		0.3500E 02	0.42915661E 00
㉟		0.3600E 02	0.44227310E 00
㉟		0.3700E 02	0.47303333E 00
㉟		0.3800E 02	0.48060189E 00
㉟		0.3900E 02	0.48610077E 00
㉟		0.4000E 02	0.49248171E 00
㉟		0.4100E 02	0.50097942E 00
㉟		0.4200E 02	0.506090910E 00
㉟		0.4300E 02	0.51010744E 00
㉟		0.4400E 02	0.51299738E 00
㉟		0.4500E 02	0.51470449E 00
㉟		0.4600E 02	0.51251504E 00
㉟		0.4700E 02	0.51353928E 00
㉟		0.4800E 02	0.51441196E 00
㉟		0.4900E 02	0.51288563E 00
㉟		0.5000E 02	0.51091964E 00

29.7.4a

Peak  
32.4% overshoot

④ NASAP ETRANSLAG APPROXIMATION

④ NONE  
STEP RESPONSE

NUMBER OF LOOPS PER ORDER

V1 1 2 1  
R8 2 3 1  
V2 1 4 1 IR3  
R1 4 5 89.7  
C1 5 1 1F  
I3 1 6 13.1 VC1  
R4 6 1 1UK  
R2 6 7 1

1= 8  
2= 19  
3= 13  
4= 1

C2 7 4 -0.475F  
R3 7 8 -2  
C3 8 1 1.425F  
R5 8 1 2  
V4 1 3 1 VR3  
V5 1 9 0.53H VR4

OUTPUT  
VV5/VV1  
TIME 50  
EXECUTE

TRANSFER FUNCTION VV5/VV1

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$$( 3.69E-01 \quad -1.05E+00 \text{ S} \quad +1.00E+00 \text{ S}^2 )$$

$$H(s) = 7.856e-02$$

$$( 5.81E-02 \quad +2.27E-01 \text{ S} \quad +1.21E+00 \text{ S} \quad +1.00E+00 \text{ S}^3 )$$

ZERO OF TRANSFER FUNCTION

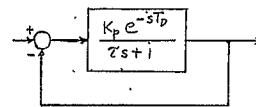
ZERO REAL PART IMAG. PART

1 C-0.26 2E 00 -0.30387E 00  
2 0.526 2E 00 0.30387E 00

POLE OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

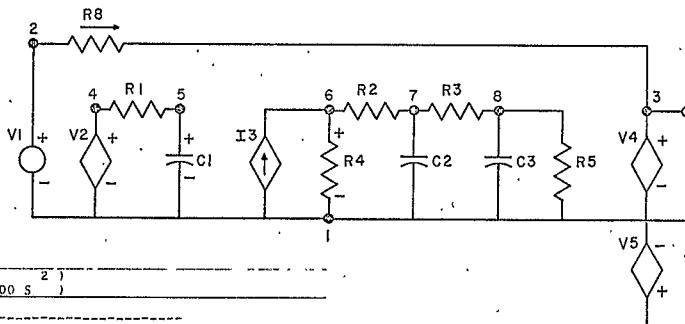
1 -0.82196E-01 0.22087E 00  
2 -0.32158E-01 -0.22087E 00  
3 -0.15452E 01 0.31018E 00



$$\zeta = 89.7$$

$$T_D = 5.7$$

$$K_p = 13.1$$



247/3

	STEP	RESPONSE
	-3.40E-01	-2.40E-01 -1.40E-01 -4.00E-02 6.00E-02 1.60E-01 2.60E-01 3.60E-01 4.60E-01 5.60E-01 6.60E-01
	0.0000E 00	.....*.....I.....*
	5.0000E 00	.....*.....I.....*
	1.0000E 01	.....*.....I.....*
	1.5000E 01	.....*.....I.....*
	2.0000E 01	.....*.....I.....*
	2.5000E 01	.....*.....I.....*
62	3.0000E 01	.....*.....I.....*
	3.5000E 01	.....*.....I.....*
	4.0000E 01	.....*.....I.....*
	4.5000E 01	.....*.....I.....*
	4.9999E 01	.....I.....*

207.46

## VIIB EXAMPLE OF A CONTROL SYSTEM WITH TRANSPORT LAG

As an extension of the concepts introduced in the previous example, we consider a given control system that has a plant with transport lag along with three simple lags shown in Fig. 7.5. Lupfer and Oglesby [LU 1] discussed such a problem wherein the object was to find a proportional-integral controller. They treated this problem with the aid of an analog computer. An alternative solution was presented by Eisenberg [EI 2] using the parameter plane approach to develop a graphical technique.

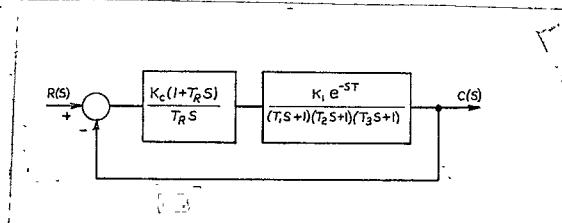


Fig. 7.5 Control system with transport lag  
Reactor process from [LU 1]

To use NASAP we must obtain a circuit model for the control system with transport lag. From the previous approximation problem we know how to handle  $e^{-sT_D}$  by using a Padé approximation that yields a rational function realized by a ladder network. For the controller of this system we require a different type of model. We can write

$$\frac{K_c(T_r s + 1)}{T_r s} = K_c \left(1 + \frac{1}{T_r s}\right) \quad (7.17)$$

Now we use the model shown in Fig. 7.6 where V7 and I2 are dependent on the error signal and I1 is dependent on the current through LL. The first three elements model the  $\frac{1}{T_r s}$  term while the fourth element accounts for the unity term. The rest of the model poses no new problems.

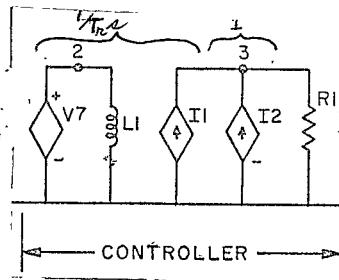


Fig. 7.6 Controller Circuit Model

To be specific then we are given the transfer function of the plant

$$G(s) = \frac{K_1 e^{-sT}}{(T_1 s + 1)(T_2 s + 1)(T_3 s + 1)} \quad (7.18)$$

where

$$K_1 = 1.0$$

$$T_1 = 13.1 \text{ min}$$

$$T_2 = 11.1 \text{ min}$$

$$T_3 = 0.5 \text{ min}$$

$$T = 9.5 \text{ min}$$

To obtain the unit step response we must determine the controller gain  $K_c$  and the reset time  $T_r$ . From [LU 1] we have the experimental results obtained with an analog computer. This is shown in Fig. 7.7 where the key response characteristics are labeled and the controller parameters used by Lupfer and Oglesby are indicated.

Following Eisenberg [EI 2] we can restate the design problem as:

Choose values of controller gain constant  $K_c$  and reset time  $T_r$  to yield an output time domain response with

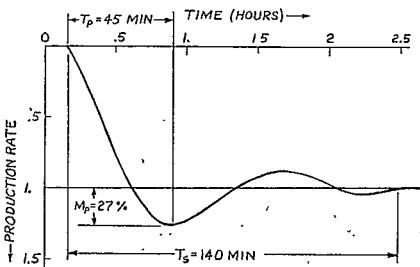


Fig. 7.7 Reactor process; response to unit step input,  $K_C = 1.0$  and  $T_r = 20.8$  min.

the characteristics of Fig. 7.7. Specifically, it is desired that (1) the peak overshoot  $M_p = 27\%$ , (2) the time to reach a peak  $T_p = 45$  min (where the output response begins after the initial  $T = 9.5$  min delay), and (3) the time for the response to reach  $2\%$  of the steady-state value  $T_s = 140$  min.

At this point we do not have to reformulate the system transfer function as in [EI 2] but simply obtain the NASAP model shown in Fig. 7.8.

Finally we obtain the unit step response using NASAP. Compare this output Fig. 7.9 with that given in Fig. 7.7. This ability to handle transport lag considerably broadens the range of aerospace control problems that can be assisted by the use of NASAP. To permit detailed comparisons Fig. 7.9a is based on  $K_C = 1$  and  $T_r = 20.8$  min. while Fig. 7.9b is a rerun of this example using the values  $K_C = 0.9$  and  $T_r = 21.9$  min. taken from [EI 2].

## ④ NASAP LUPPER-UGLESBY CONTROL PROCESS

11SM-N1311

⑤ NONE  
STEP RESPONSE4/14/70  
43.0270 SEC.

NUMBER OF LOOPS PER ORDER

V1 1 10 1	1=	12
R6 10 11 1	2=	40
V7 1 2 1 IR6	3=	47
I1 2 1 1249H	4=	26
I2 1 3 1 IR6	5=	9
I1 1 3 1 IL1	6=	1
R1 3 1 1		
V2 1 4 1 VR1		
R2 4 5 786		
C2 5 1 1E		
V3 1 6 1 VC2		
R3 6 7 666		
C3 7 1 1F		
V4 1 8 1 VC3		
R4 8 9 30		
C4 9 1 1E		
I5 1 12 1 VC4		
R7 12 1 LOK		
R5 12 13 1		
C5 13 1 -47.5F		
R9 13 14 -2		
C9 14 1 142.5F		
R8 14 1 2		
V6 1 11 1 VR7		
OUTPUT VV6/VV1		
VV6EY9000		
TIME=9000		
EXECUTE		

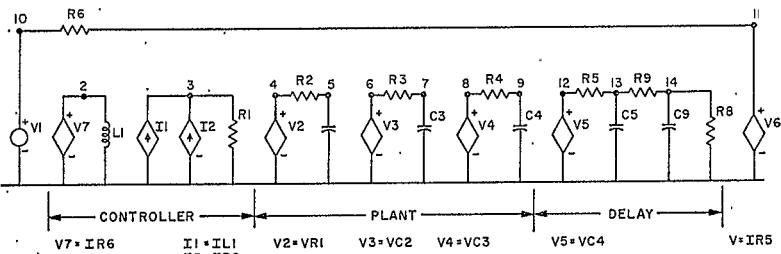


Fig. 7.8 NASAP Model for Fig. 7-5

$$\text{⑥ TRANSFER FUNCTION } VV6/VV1 \\ \text{ } \\ \text{ } \quad ( 2.96E-08 \quad +2.85E-05 s \quad -9.73E-03 s^2 \quad +1.00E-00 s^3 )$$

$$\text{⑦ } H(s) = 6.367E-08 s$$

$$\text{⑧ } ( 1.88E-15 \quad +4.17E-12 s \quad +3.54E-09 s^2 \quad +2.45E-06 s^3 \quad +5.11E-04 s^4 \quad +4.66E-02 s^5 \quad +1.00E-00 s^6 )$$

⑨

ZERO OF TRANSFER FUNCTION

POLE OF TRANSFER FUNCTION

⑩ ZERO REAL PART IMAG. PART

POLE REAL PART IMAG. PART

$$\begin{array}{ll} 1 & 0.52632E-02 \quad 0.30387E-02 \\ 2 & 0.52632E-02 \quad -0.30387E-02 \\ 3 & -0.60128E-03 \quad 0.00000E 00 \end{array}$$

$$\begin{array}{ll} 1 & -0.58469E-02 \quad 0.40666E-02 \\ 2 & -0.58469E-02 \quad -0.40666E-02 \\ 3 & -0.68800E-03 \quad 0.00000E 00 \\ 4 & -0.40142E-03 \quad -0.12052E-02 \\ 5 & -0.40134E-03 \quad 0.12052E-02 \\ 6 & -0.33447E-01 \quad 0.00000E 00 \end{array}$$

STEP	RESPONSE FUNCTION	TIME	STEP	RESPONSE
				VVB/VV1
F(T) =		0.0000E 00		-0.8940696E-06
	(-0.1259E 00 J -0.6185E-01 ) E	0.1800E 03		0.50302744E-02
	(-0.5847E-02 J 0.4067E-02 ) T	0.3600E 03		-0.27541515E-02
		0.5400E 03		-0.79890554E-02
	(-0.1059E 00 J -0.6185E-01 ) E	0.7200E 03		0.1506900E-01
	(-0.5847E-02 J -0.4067E-02 ) T	0.9000E 03		0.73874056E-01
		0.1080E 04		0.16472322E 00
	(-0.6880E-03 J 0.0000E 00 ) T	0.1260E 04		0.27963656E 00
	(-0.2157E 00 J -0.3780E-07 ) E	0.1440E 04		0.41066960E 00
		0.1620E 04		0.56791850E 00
	(-0.4013E-03 J -0.1206E-02 ) T	0.1800E 04		0.68588519E 00
	(-0.2846E 00 J -0.5080E 00 ) E	0.1980E 04		0.81753464E 00
		0.2160E 04		0.93795317E 00
	(-0.4013E-03 J 0.1206E-02 ) T	0.2340E 04		0.10428429E 01
	(-0.2846E 00 J -0.5080E 00 ) E	0.2520E 04		0.11295256E 01
		0.2700E 04		0.11953567E 01
	(-0.3345E-01 J 0.0000E-09 ) T	0.2880E 04		0.12434797E 01
	(-0.3361E-02 J 0.2240E-12 ) E	0.3060E 04		0.12709026E 01
		0.3240E 04		0.12802896E 01
	( 0.0000E 00 J 0.0000E 00 ) T	0.3420E 04		0.12737522E 01
	( 0.1000E 01 J 0.3203E-06 ) E	0.3600E 04		0.12530738E 01
		0.3780E 04		0.12235241E 01
		0.3960E 04		0.11956804E 01
		0.4140E 04		0.11432724E 01
		0.4320E 04		0.10994045E 01
		0.4500E 04		0.10554561E 01
		0.4680E 04		0.10146941E 01
		0.4860E 04		0.97612319E 00
		0.5040E 04		0.94726765E 00
		0.5220E 04		0.92280972E 00
		0.5400E 04		0.90511072E 00
		0.5580E 04		0.88941515E 00
		0.5760E 04		0.86985850E 00
		0.5940E 04		0.85079757E 00
		0.6120E 04		0.82967752E 00
		0.6300E 04		0.80716964E 00
		0.6480E 04		0.92036349E 00
		0.6660E 04		0.93552864E 00
		0.6840E 04		0.95167798E 00
		0.7020E 04		0.96790797E 00
		0.7200E 04		0.98343128E 00
		0.7380E 04		0.99759942E 00
		0.7560E 04		0.10099154E 01
		0.7740E 04		0.10200396E 01
		0.7920E 04		0.10277815E 01
		0.8100E 04		0.10330915E 01
		0.8280E 04		0.10360460E 01
		0.8460E 04		0.10368204E 01
		0.8640E 04		0.10356770E 01
		0.8820E 04		0.10329304E 01
		0.9000E 04		0.10209335E 01

	STEP	RESPONSE
	-7.20E-01	-5.20E-01
	-3.20E-01	-1.20E-01
	8.00E-02	2.80E-01
	4.80E-01	6.80E-01
	8.80E-01	8.80E-01
	1.08E 00	1.28E 00
0.0000E_00	*	*
	*	*
	*	*
	*	*
9.0000E_02	*	*
	*	*
	*	*
	*	*
1.8000E_03	*	*
	*	*
	*	*
	*	*
2.7000E_03	*	*
	*	*
	*	*
	*	*
3.6000E_03	*	*
	*	*
	*	*
	*	*
4.5000E_03	*	*
	*	*
	*	*
	*	*
5.4000E_03	*	*
	*	*
	*	*
	*	*
6.2999E_03	*	*
	*	*
	*	*
	*	*
7.1999E_03	*	*
	*	*
	*	*
	*	*
8.0999E_03	*	*
	*	*
	*	*
	*	*
8.9999E_03	*	*
	*	*
	*	*
	*	*

ASM-IV.H5AP

4/20/70

39.7947 SEC.

⑩ NASAP LUPFER-UGLESBY CONTROL PROCESS

REPEATED USING EISENBERG VALUES [EII 2]

$K_C = 0.9$

⑪ NONE  
STEP RESPONSE

$T_R = 21.9$

⑫ V1 1 10 1

R6 10 11 1

V7 1 2 1 IR6

L1 2 1 1 1214H

I2 1 3 1 IR6

I1 1 3 1 IL1

R1 2 1 0.9

V2 1 4 1 VR1

R2 4 5 706

C2 5 1 1F

V3 1 6 1 VC2

R3 6 7 666

C3 7 1 1F

NUMBER OF LOOPS PER ORDER

1= 11

2= 30

3= 26

4= 13

5= 3

V4 1 8 1 VC3

R4 8 9 30

C4 9 1 1F

V5 1 12 1 VC4

R5 12 13 1

C5 13 1 47.5F

P9 13 1 4 -2

C9 14 1 -142.5F

R8 14 1 2

V6 1 11 1 IR5

⑬ OUTPUT

VV6/VV1

TIME 9000

⑭ EXECUTE

⑮ TRANSFER FUNCTION VV6/VV1

( 2.81E-08 +2.89E-05 S -9.77E-03 S <sup>2</sup> +1.00E 00 S <sup>3</sup> )

⑯ H(S) = 5.731E-08 \*-----

( 1.61E-15 +4.01E-12 S +3.60E-09 S <sup>2</sup> +2.45E-06 S +5.11E-04 S +4.66E-02 S +1.00E 00 S <sup>5</sup> )

⑰ POLE OF TRANSFER FUNCTION

⑱ ZERO OF TRANSFER FUNCTION

POLE REAL PART IMAG. PART

⑲ ZERO REAL PART IMAG. PART

1 -0.52622E-02 -0.30347E-02

2 -0.52622E-02 -0.30347E-02

3 -0.76104E-03 0.00000E 00

1 -0.58147E-02 0.39870E-02

2 -0.58147E-02 -0.39870E-02

3 -0.47543E-03 -0.11591E-02

4 -0.47543E-03 0.11591E-02

5 -0.61752E-03 0.00000E 00

39.7947 !

STEP	RESPONSE FUNCTION	STDP	TIME	RESPONSE
1	$F(T) =$		0.0000E 00	VV6/VV1 -0.95367432E-06
2	(-0.5815E-02 J 0.5839E-01) E		0.1800E 03	0.45130849E-06
3	(-0.1C12E 00 J 0.5839E-01) E		0.3600E 03	-0.29047488E-02
4	(-0.5815E-02 J-0.5839E-01) E		0.5400E 03	-0.71744919E-02
5	(-0.1012E 00 J-0.5839E-01) E		0.7200E 03	0.13525119E-01
6	(-0.4754E-03 J-0.1159E-02) T		0.9000E 03	0.66282332E-01
7	(-0.2453E 00 J-0.5323E 00) E		0.1080E 04	0.14736704E 00
8	(-0.4754E-03 J 0.1159E-02) T		0.1260E 04	0.24952722E 00
9	(-0.2453E 00 J 0.5323E 00) E		0.1440E 04	0.36511564E 00
10	(-0.4754E-03 J 0.1159E-02) T		0.1620E 04	0.48707902E 00
11	(-0.2453E 00 J 0.5323E 00) E		0.1800E 04	0.60921687E 00
12	(-0.6175E-03 J 0.0000E 00) T		0.1980E 04	0.72628343E 00
13	(-0.3038E 00 J-0.1050E-06) E		0.2160E 04	0.83405477E 00
14	(-0.3038E 00 J-0.1050E-06) E		0.2340E 04	0.92936039E 00
15	(-0.3344E-01 J 0.0000E 00) T		0.2520E 04	0.10100660E 01
16	(-0.3535E-02 J 0.7430E-10) E		0.2700E 04	0.10750170E 01
17	(-0.3344E-01 J 0.0000E 00) T		0.2880E 04	0.11239424E 01
18	(-0.3535E-02 J 0.7430E-10) E		0.3060E 04	0.11573285E 01
19	(-0.3344E-01 J 0.0000E 00) T		0.3240E 04	0.11762687E 01
20	(-0.3535E-02 J 0.7430E-10) E		0.3420E 04	0.11829590E 01
21	(-0.3344E-01 J 0.0000E 00) T		0.3600E 04	0.11775694E 01
22	(-0.3535E-02 J 0.7430E-10) E		0.3780E 04	0.11639090E 01
23			0.3960E 04	peak at 18.2% - 57 min. (470 sec.)
24			0.4140E 04	0.11435795E 01
25			0.4320E 04	0.11187010E 01
26			0.4500E 04	0.10912590E 01
27			0.4680E 04	0.10620390E 01
28			0.4860E 04	0.10355569E 01
29			0.5040E 04	0.10106063E 01
30			0.5220E 04	0.98749797L 00
31			0.5400E 04	0.95350277L 00
32			0.5580E 04	0.94257486E 00
33			0.5760E 04	0.93564713E 00
34			0.5940E 04	0.93255331E 00
35			0.6120E 04	0.93259393L 00
36			0.6300E 04	0.93567178E 00
37			0.6480E 04	0.94082105E 00
38			0.6660E 04	0.94777817E 00
39			0.6840E 04	0.95587045E 00
40			0.7020E 04	0.96456379E 00
41			0.7200E 04	0.9737794E 00
42			0.7380E 04	0.98190094E 00
43			0.7560E 04	0.98979634E 00
44			0.7740E 04	0.99580966E 00
45			0.7920E 04	0.10027647E 01
46			0.8100E 04	0.10075626E 01
47			0.8280E 04	0.10111713E 01
48			0.8460E 04	0.10146165E 01
49			0.8640E 04	0.10149803E 01
50			0.8820E 04	0.10153723E 01
51			0.9000E 04	0.10149336E 01

4474

	STEP	RESPONSE
	-8.20E-01 -6.20E-01 -4.20E-01 -2.20E-01 -2.00E-02 1.80E-01 3.80E-01 5.80E-01 7.80E-01 9.80E-01 1.18E-00	
0.0000E 00	.....	.....
	*	*
	*	*
	*	*
	*	*
9.0000E 02	.....	.....
	*	*
	*	*
	*	*
1.8000E 03	.....	.....
	*	*
	*	*
	*	*
2.7000E 03	.....	.....
	*	*
	*	*
	*	*
3.6000E 03	.....	.....
	*	*
	*	*
	*	*
4.5000E 03	.....	.....
	*	*
	*	*
	*	*
5.4000E 03	.....	.....
	*	*
	*	*
	*	*
6.2999E 03	.....	.....
	*	*
	*	*
	*	*
7.1999E 03	.....	.....
	*	*
	*	*
	*	*
8.0999E 03	.....	.....
	*	*
	*	*
	*	*
8.9999E 03	.....	.....
	*	*
	*	*
	*	*

#### VIII. EXAMPLE SHOWING NASAP LIMITATION

We now present an active filter circuit which illustrates a flowgraph of great complexity and demonstrates some NASAP limitations. In particular the fact that it may not be sufficient that the user be judicious in the specific tree he allows the NASAP program to select. In Fig. 7.9 which is identical to Fig. 2.16 in [HU 1] is shown the 28-element NASAP circuit diagram of this Hutton problem. The input voltage  $V_1$  is fed through a low-pass tee-network to the base of a transistor which is part of a two-transistor differential amplifier. The output voltage of this differential amplifier is connected directly to the base of a simple common emitter transistor stage. The voltage at the collector of this transistor, which is also the output voltage of the circuit, is fed back through an RC twin-tee network to the base of the other transistor in the differential amplifier. Each of these three transistors is represented in Fig. 7.9b by the h-parameter equivalent circuit with  $h_{12} = 0$  and with a capacitor ( $C_2, C_3, C_7$ ) connected between the base and collector terminals. This capacitance is included to take into account the frequency characteristic of the transistor.

The NASAP input listing used by Hutton is reproduced as Fig. 7.10. There all 17 of the resistors are listed first and followed by the seven capacitors, in numerical sequence. With this listing, the NASAP tree selection algorithm selects as branches of the tree the elements in the following order;  $V_1, R17, C1, C2, C3, C4, C6, C7, R4, R5, R8, R15$ . This particular tree generates a flowgraph possessing a total number of loops of all orders of 2,440,105 a very complicated flowgraph indeed. As noted by Hutton, 10 minutes of execution time were required on the UNIVAC 1108.

Utilizing one of the options discussed in Chapter II, a tree can be selected to yield a flowgraph with considerably fewer loops. Seven of the

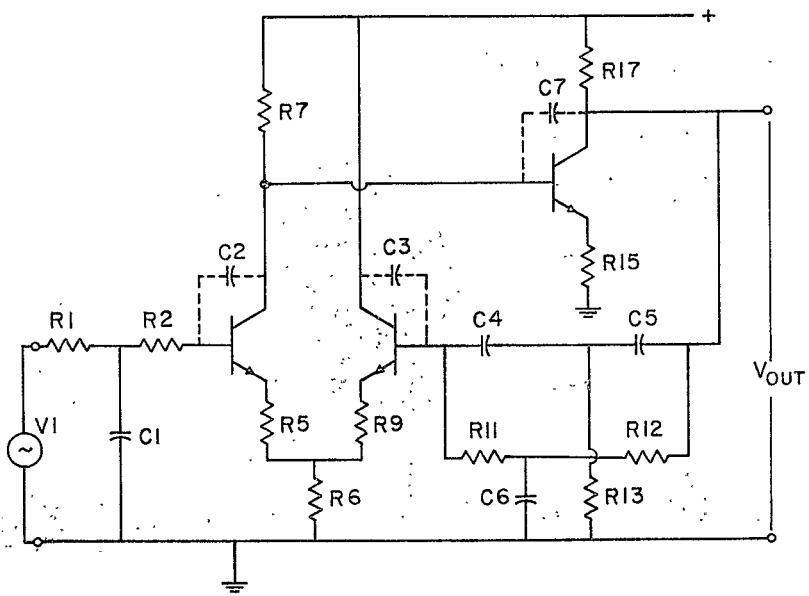


Fig. 7.9a Active Filter Circuit for Hutton Problem (Circuit 6)

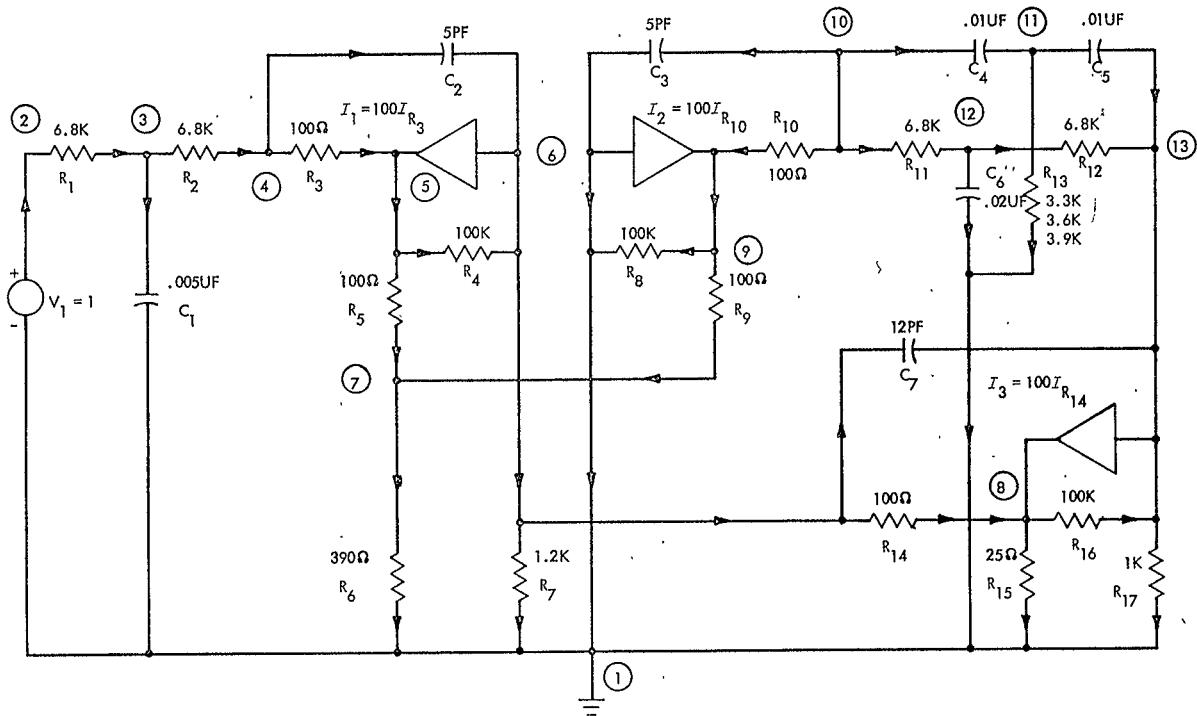


Fig. 7.9b Hutton Problem

NASAP

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30

R1 2 3 6.8K  
 R2 3 4 6.8K  
 R3 4 5 100  
 R4 5 6 100K  
 R5 6 7 100  
 R6 7 1 390  
 R7 6 1 1.2K  
 R8 9 1 100K  
 R9 9 7 100  
 R10 10 8 100  
 R11 10 12 6.8K  
 R12 12 13 6.8K  
 R13 11 1 3.6K  
 R14 6 8 100  
 R15 8 1 25  
 R16 8 15 100K  
 R17 13 1 1K  
 C1 3 1 .5UF  
 C2 4 6 .005UF  
 C3 10 1 .005UF  
 C4 10 11 100F  
 C5 11 13 100F  
 C6 12 1 200F  
 C7 6 13 .012UF  
 V1 1 2 1.  
 I1 6 5 100 IR3  
 I2 1 9 100 IR10  
 I3 13 8 100 IR14  
 OUTPUT  
 VR17/VV1  
 FREQ -1 4 .1  
 TIME .005  
 EXECUTE

32

## NUMBER OF LOOPS PER ORDER

1=	696.	2 20
2=	8969	
3=	52922	
4=	185371	
5=	419756	
6=	628027	
7=	616324	
8=	380754	
9=	138488	
10=	26132	
11=	.1896..	

TRANSFER FUNCTION VR17/VV1

$$F(S) = \frac{.666+06s^1 + (.16+17s^2 - (.49+16s^3 - (.33+15s^4 - (.44+13s^5 - (.43+07s^6 + (.10+01s^7))}{.11+23s^1 + (.69+21s^2 - (.68+20s^3 - (.60+19s^4 - (.11+18s^5 - (.38+13s^6 + (.91+07s^7))})$$

Fig. 7.10 NASAP Printout for Hutton Circuit 6

elements (namely, V1, R17, C1, C3, C6, R8, R15) in the above described tree are connected to the node numbered 1 in Fig. 7.11. However, three resistors, R6, R7, and R13 are also connected to node 1. If these three elements can be made branches of a tree which also contains the seven elements from the tree selected by Hutton, then ten of the twelve necessary tree branches will be connected to a common node. Such a tree with a definite star-like structure should yield a flowgraph with fewer loops. We shall now indicate how R6, R7, and R13 can be selected as tree branches and which branches of the original Hutton tree must be removed to make way for these resistors.

The resistor R7 forms a closed path with the original tree branches C7 and R17. Since the voltage across R17 is the specified output variable, R7 will become a tree branch only if C7 can be removed from the tree. This is easily accomplished by adding a dummy voltage source dependent on the voltage across R7. Likewise the resistor R13 forms a closed path with the original tree branches C3 and C4. Since C3 is connected to node 1, we wish to make R13 a tree branch in place of C4. A dummy voltage source dependent upon the voltage across R13 easily accomplishes this by making R13 a type 2 element instead of a type 4 element (see description of tree selection algorithm). Finally R6 can be included in the list of tree branches by making it the first resistor described in the NASAP input listing. With R6 in the tree, either R4 or R5 must be removed from the tree. The choice is easily made by noting that the co-tree element R3 will form a closed path with R4 and C2 when R4 is a tree branch. On the other hand, with R5 as a tree branch, R3 will form a closed path with R6, R7, R5 and C2. Hence R4 should remain as a tree branch. This is achieved by having R4 precede R5 in the NASAP input list.

The revised NASAP input listing that yields a tree with 10 elements connected to the same node is shown in Fig. 7.12. The controlled sources V2

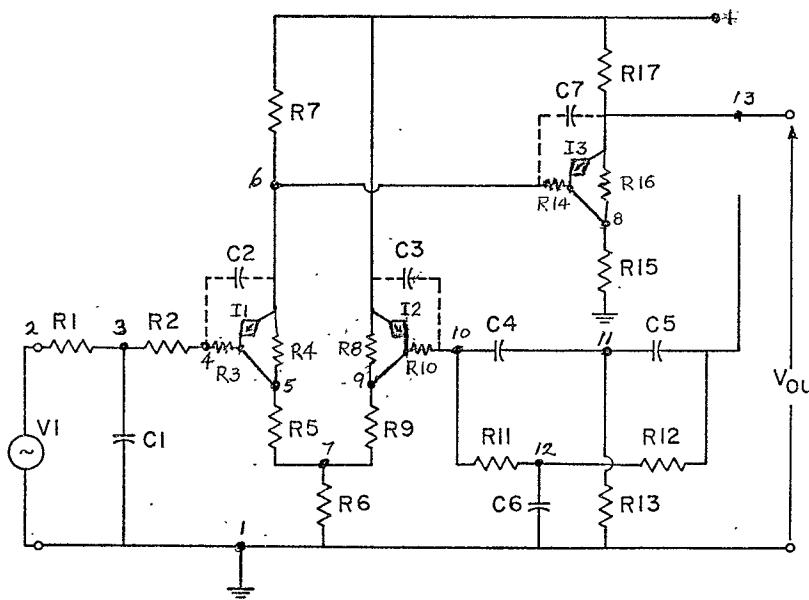


Fig. 7.11 Alternative NASAP Model for Hutton  
(compare Fig. 7.9)

NASAP PPC3LC 5 FR... ITI.

None  
None

V1 1 2 1.  
R6 7 1 390  
R15 8 1 25  
R1 2 3 6.31  
R2 3 4 6.81  
R3 4 5 100  
R4 5 6 170K  
R5 7 8 100  
R7 9 1 1.2K  
R8 9 1 100  
R9 9 7 100  
R10 10 9 120  
R11 10 12 6.8  
R12 12 11 6.8  
R13 13 1 3.11  
R14 6.8 10  
R16 9 11 1.1K  
R17 11 1 1K  
C1 3 1 2uF  
C2 4 6.0uF  
C3 10 1 .01F  
C4 10 13 16  
C5 13 11 1  
C6 12 1 20K  
C7 5 11 1  
I1 6.5 120 1.3  
I2 1 9 14.0 1.1  
I3 11 8 100 1.14  
V2 1 14 1.0 1.7  
V3 1 15 1.0 1.13  
OUTPUT  
VR17/VV1  
EXECUTE

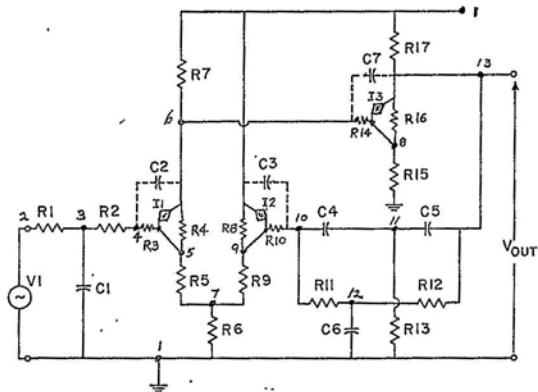


Fig. 7.12

ASM-NASAP

3/11/70

25 min = 1509.2789 SEC

DUMP in MULLER

(102-415,515) long

11/11/70

and  $V_3$  are the dummy voltage sources necessary to include  $R_7$  and  $R_{13}$  in the tree selected by NASAP. With these 2 elements included, there are 30 elements in the equivalent circuit (NOTE: 30 elements are the maximum number of elements that can be used on the RCA Spectra 70, and IBM 360 machines because of their 32 bit computer words). Note in Fig. 7.12 that the flowgraph generated by this new tree contains 415, 515 loops of all orders. Thus a saving of over 2 million loops has been achieved by careful selection of the tree. The subsequent execution time was 25 minutes on the Spectra 70/46 (equivalent to approximately 3 minutes on the IBM 360/75). Note also the difference between the transfer function found by use of the original Hutton tree (Fig. 7.10) and that found by use of the tree described here (Fig. 7.12). The extra 2 million loops results in considerable error in the coefficients of the transfer function, see the discussion in [SE 1]. Note in Fig. 7.12 that only the zeros of the transfer function are given. Due to excessive floating point overflow, the MULLER (root-finding) subroutine was unable to determine the poles of the transfer function.

Through the cooperation of Prof. Alan B. Macnee the Hutton problem was run using CIRAN (a program based on state variables) on the University of Michigan IBM 360/67 computer. The impedances were scaled by  $10^{-3}$  and frequency  $10^{-6}$ . The circuit was found to exhibit a pair of dominant conjugate complex poles near 2.4 KHz. For a  $\pm 10\%$  change in the value of  $R_{13}$  the Q of this pole ranged from 7.5 to 20. The fact that these three analyses took only 12.13 seconds of CPU time indicates a severe limitation of NASAP for this class of problems.

In Fig. 7.13 the compensation is seen to consist of two identical RC lag network. The transfer function of the RC lag network is

$$\frac{\frac{1}{RC}}{S + \frac{1}{RC}} \quad \text{or} \quad \frac{G}{S + G}$$

The parameter  $G$ , the RC lag network pole, is chosen different from all plant poles, usually farther inside the left half plane than any pole of the desired system.

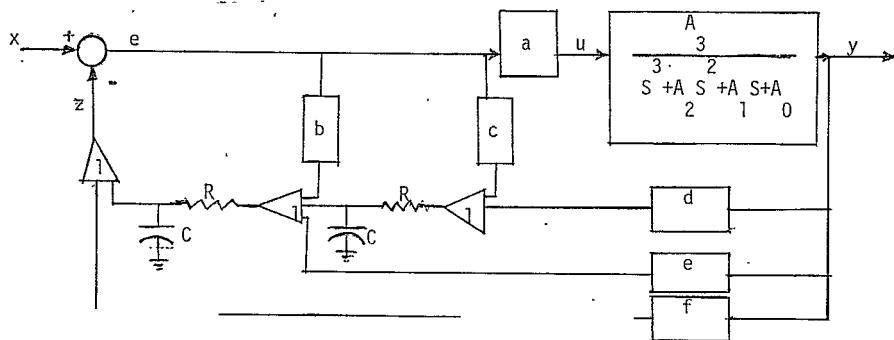


Fig. 7.13 Compensated Third Order Control System

Fig. 7.14 is equivalent to Fig. 7.13 with respect to the transfer function involved.

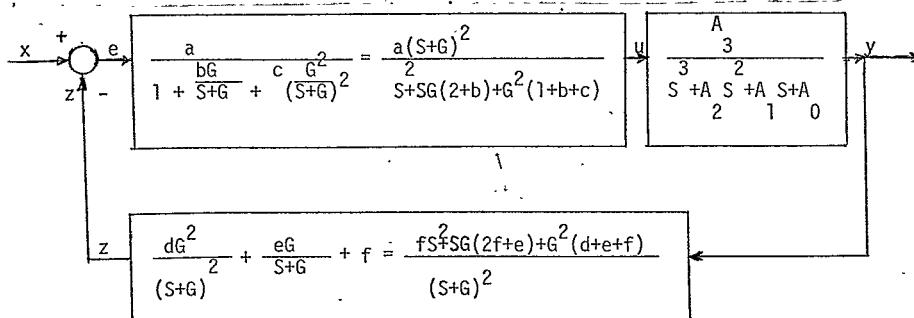


Fig. 7.14 Equivalent Feedback Control System Representation for Fig. 7.13

The closed-loop transfer function for  $y/x$  shown in Fig. 7.14 is

$$\frac{y}{x} = \frac{a(s+G)^2 A_3}{1 + \frac{(s^2 + SG(2+b) + G^2(1+b+c))(s^3 + A_2 s^2 + A_1 s + A_0)}{(s^2 + SG(2+b) + G^2(1+b+c))(s^3 + A_2 s^2 + A_1 s + A_0)}} \quad (7.21)$$

Equation (7.20) for the desired system, when multiplied top and bottom by  $(s+G)^2$ , becomes

$$\frac{y}{x} = \frac{B_3(s+G)^2}{s^5 + (B_2 + 2G)s^4 + (B_1 + 2GB_2 + G^2)s^3 + (B_0 + 2GB_1 + G^2B_2)s^2 + (G^2B_1 + 2GB_0)s + G^2B_0} \quad (7.22)$$

This equation can be compared termwise with (7.21) after it has been rewritten to clear fractions. Thus we find that  $aA_3 = B_3$  and successively

$$b = \frac{(B_2 - A_2)}{G} \quad \text{where } G \text{ is positive.}$$

$$c = \frac{(B_1 - A_1) + (G - A_2)(B_2 - A_2)}{G^2} \quad \text{with } G \text{ positive.}$$

$$f = \frac{(B_0 - A_0) + (B_1 - A_1)(2G - A_2) + (B_2 - A_2)((G - A_2)^2 - A_1)}{B_3}, \quad B_3 \neq 0$$

$$c = \frac{(B_1 - A_1)(-3G^2 - A_1 + 2GA_2) + (B_2 - A_2)(-2G(G - A_2)^2 + A_1 A_2 - A_0)}{GB_3}$$

and finally

$$d = \frac{(B_1 - A_1)(G^3 - G^2 A_2 + GA_1 - A_0) + (B_2 - A_2)(G^2 A_1 + A_0 A_2 - GA_0 + G^2(G - A_2)^2 - GA_1 A_2)}{G^2 B_3} \quad (7.23)$$

#### VIID EXAMPLE INVOLVING LUENBERGER OBSERVER

For the last problem we consider an application of modern control theory based on state variables to show the versatility of NASAP. In particular we use the Luenberger observer method to implement compensation for a control system situation in which all the states of the system are not measurable. For present purposes it is sufficient to note that the Luenberger observer [LU 1] is a device which constructs an estimate of the system state vector based upon the available system inputs and outputs. Then, based upon the reconstructed state vector, simple matrix algebra manipulations can be utilized to obtain estimates of the missing states or a combination of the missing states.

Luenberger has shown that for an n-th order system with m measureable states, the order of the required observed need only contain n-m poles. Furthermore these pole locations are arbitrary as long as they are different from the eigenvalues of the system matrix.

The Luenberger observer accomplishes the desired result by adding dynamics in the feedback path of the control system.

This theory suggests a unique form for general compensation of third order systems wherein the designer can place the closed-loop poles at any desired location. This development is adapted from Newman [NE 2].

The open loop descriptions of the control system shown in Fig. 7.13 is

$$\frac{y}{u} = \frac{A_3}{s^3 + A_2 s^2 + A_1 s + A_0} \quad (7.19)$$

For convenience the desired closed-loop system transfer function is expressed as

$$\frac{y}{x} = \frac{B_3}{s^3 + B_2 s^2 + B_1 s + B_0} \quad (7.20)$$

These six equations define unique values for the symbols a, b, c, f, e, and d respectively.

It is worth noting [NE 2] that this method of general system compensation may be extended to higher order systems. An N-th order system will require a string of N-1 RC lag networks. Each lag network is driven by the signal e and the signal y, both passing through gain blocks as in Fig. 7.13. The z signal is formed by adding the output of the string and a signal equal to fy, as in Fig. 7.13. The a block is used in cascade with the plant as in Fig. 7.13 .

It is found easier to design compensators for plant transfer functions with numerator polynomials by reducing the coefficients in the equations to numbers, instead of trying to derive the general relationship.

For an illustrative example, we consider the open loop and desired closed loop transfer functions for the control system in Fig. 7.13 .

#### Open Loop System

$$\frac{y}{u} = \frac{A_3}{s^3 + A_2 s^2 + A_1 s + A_0} = \frac{10}{s(s+1)(s+10)} \quad (7.24)$$

#### Closed Loop System (Desired)

$$\frac{y}{x} = \frac{B_3}{s^3 + B_2 s^2 + B_1 s + B_0} = \left( \frac{5}{s+5} \right) \left( \frac{20}{s+20} \right)^2 \quad (7.25)$$

Following the procedure outlined above, we determine the constants a, b, c, d, e, f, as listed.

$a = 200$	$b = 2.24$
$c = 1.7$	$d = 4.256$
$e = -14.018$	$f = 10.762$

(7.26)

The parameter  $G$ , was chosen as twenty ( $G = 20$ ) which is different from all plant poles.

Having thus specified the Luenberger observer we need the corresponding electric circuit model for the compensated third order control system. This model is shown in Fig. 7.15. A NASA run was made for the step response of this model. We note that the transfer function shows six critical frequencies near  $s = -20$ ; two zeros and four poles. The step response shows a slight steady state error.

Execution time on the RCA Spectra 70/46 was 42.86 seconds. A comparison run was made of this third-order compensator system using CSMP (Continuous System Modeling Program) on the IBM 360/75. The CSMP step response for this control system checked very closely except that the execution time was 32 seconds (22 seconds CPU). This represents a significant cost advantage for NASA since the 360/75 is faster by a factor of approximately 8 over the Spectra 70/46. The NASA printout is shown in Fig. 7.16.

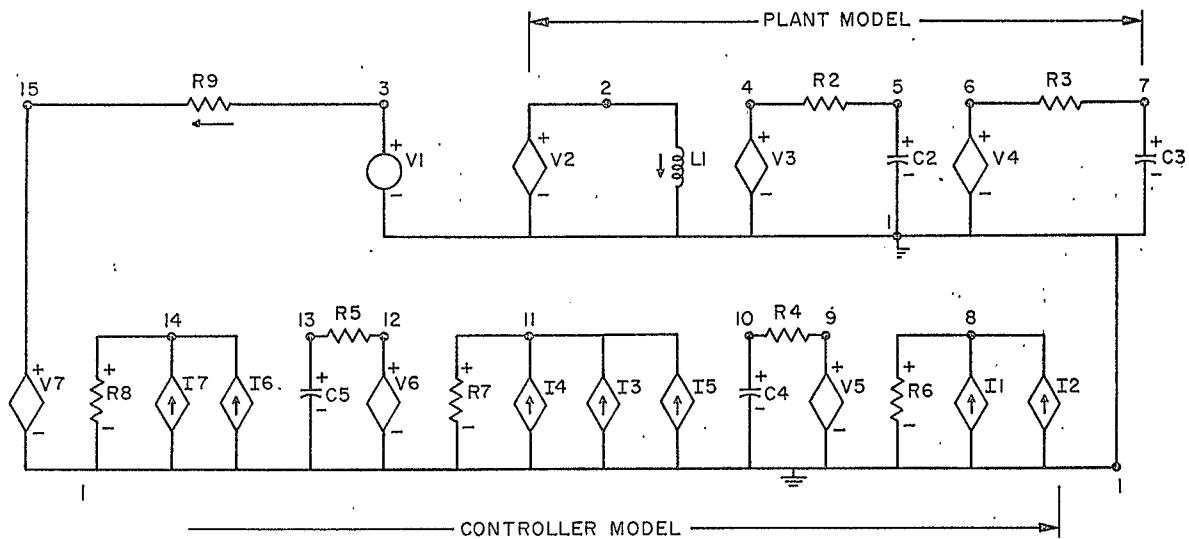


Fig. 7.15 NASAP Model for Third Order Control System in Fig. 7.13

## NASAP NEWTON DESIGN PROBLEM

1/19/70

NONE

STEP RESPONSE

V1-1 3-1  
 V2 1 2 200 IR9  
 L1 2 1 1H  
 V3 1 4 1 IL1  
 R2 4 5 1  
 C2 5 1 1F  
 V4 1-6-1 VC2  
 R3 6 7 0.1  
 C3 7 1 1F  
 I1 1 B 2.24 IR9  
 I2 1 B 4.256 VC3  
 R6 8 1 1  
 V5 1 9.1 VR6  
 P4 9 10 0.05  
 C4 10 1 1F  
 I3 1 11 1 VC4  
 I4 1-11 1.7 IR9  
 I5 1 11 -14.018 VC3  
 R7 11 1 1  
 V6 1 12 1 VR7  
 R5 12 13 0.05  
 C5 13 1 1F  
 I6 1 14 1 VC5  
 I7 1 14 10.762 VC3  
 R8 14 1 1  
 V7 1 15 1 VR8  
 R9 3 15 1

NUMBER\_OF\_LOOPS\_PER\_ORDER

OUTPUT

VC3/VV1

1= 10

TIME 1.0

2= 16

EXECUTE

3= 10

4= 2

TRANSFER FUNCTION: VC3/VV1

2:

$$( 4.00E 02 \quad +4.00E 01 S \quad +1.00E 00 S )$$

$$H(S) = 2.000E 03 \times$$

1:

$$( -8.00E -05 \quad +3.20E -05 S \quad +6.40E -05 S \quad +2.80E -03 S \quad +8.50E -01 S \quad +1.00E 00 S )$$

2:

POLE\_OF\_TRANSFER\_FUNCTION

3:

POLE REAL PART IMAG. PART

ZERO\_OF\_TRANSFER\_FUNCTION

ZERO

REAL PART IMAG. PART

$$1 -0.20692E 02 -0.69237E 00$$

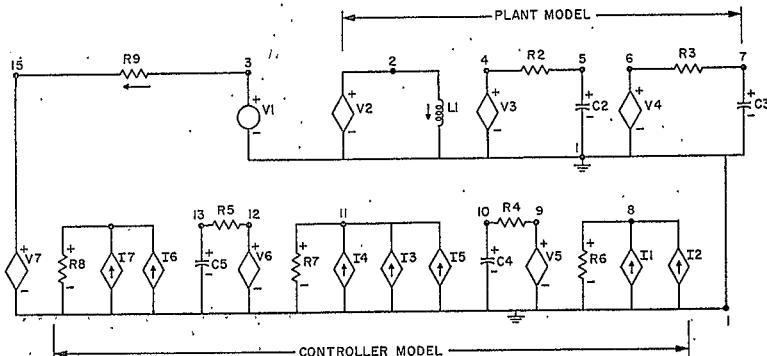
$$2 -0.20692E 02 0.69237E 00$$

$$3 -0.19325E 02 0.66244E 00$$

$$4 -0.19325E 02 -0.66244E 00$$

$$L=-0.20000E -02 -0.16526E -1$$

$$5 -0.50000E 01 0.00000E 00$$



## STEP RESPONSE FUNCTION

## STEP.. RESPONSE

R(T)

244

		TIME	VCO/VV1
		0.000E 00	0.26226044E-05
		0.200E-01	0.2356415E-02
	(-0.2062E 02 J 0.6924E 00 ) T	0.400E-01	0.13779249E-01
	(-0.1041E 01 J 0.1232E 01 ) E	0.600E-01	0.37124972E-01
	(-0.2068E 02 J 0.6924E 00 ) T	0.800E-01	0.73513739E-01
	(-0.1041E -01 J=0.1232E-01 ) E	0.100E 00	0.11717759E 00
	(-0.1932E 02 J 0.6624E 00 ) T	0.120E 00	0.16741729E 00
	(-0.1430E 01 J -0.1204E 01 ) E	0.140E 00	0.22115954E 00
	(-0.1932E 02 J -0.6624E 00 ) T	0.160E 00	0.27627476E 00
	(-0.1430E -01 J -0.1204E -01 ) E	0.180E 00	0.33111735E 00
	(-0.1932E 02 J -0.6624E 00 ) T	0.200E 00	0.38449572E 00
	(-0.1430E -01 J -0.1204E -01 ) E	0.220E 00	0.43551955E 00
	(-0.5000E -01 J 0.0000E 00 ) T	0.240E 00	0.48304955E 00
	(-0.1776E 01 J 0.4791E -08 ) E	0.260E 00	0.52209475E 00
	(-0.5000E -01 J 0.0000E 00 ) T	0.280E 00	0.57109714E 00
	(-0.1776E 01 J 0.4791E -08 ) E	0.300E 00	0.60969477E 00
	( 0.0000E 00 J 0.0000E 00 ) T	0.320E 00	0.64556819E 00
	( 0.9994E -00 J =0.1198E -08 ) E	0.340E 00	0.67825329E 00
		0.360E 00	0.70811856E 00
		0.380E 00	0.73534657E 00
		0.400E 00	0.76612996E 00
		0.420E 00	0.78265649E 00
		0.440E 00	0.80311249E 00
		0.460E 00	0.82167155E 00
		0.480E 00	0.83951956E 00
		0.500E 00	0.85375335E 00
		0.520E 00	0.87571155E 00
		0.540E 00	0.89064666E 00
		0.560E 00	0.89141941E 00
		0.580E 00	0.90167959E 00
		0.600E 00	0.91096745E 00
		0.620E 00	0.91937453E 00
		0.640E 00	0.92698336E 00
		0.660E 00	0.93386942E 00
		0.680E 00	0.94010199E 00
		0.700E 00	0.9457432E 00
		0.720E 00	0.95084363E 00
		0.740E 00	0.95546150E 00
		0.760E 00	0.95954049E 00
		0.780E 00	0.96342123E 00
		0.800E 00	0.96634259E 00
		0.820E 00	0.96993316E 00
		0.840E 00	0.97273912E 00
		0.860E 00	0.97527456E 00
		0.880E 00	0.97756815E 00
		0.900E 00	0.97964362E 00
		0.920E 00	0.98152131E 00
		0.940E 00	0.98322052E 00
		0.960E 00	0.98475796E 00
		0.980E 00	0.98614913E 00
	0.100E 01	0.98740742E 00	

	STEP	RESPONSE
	-1.00E-02	9.00E-02
	1.00E-01	1.90E-01
	2.00E-01	2.90E-01
	3.00E-01	3.90E-01
	4.00E-01	4.90E-01
	5.00E-01	5.90E-01
	6.00E-01	6.90E-01
	7.00E-01	7.90E-01
	8.00E-01	8.90E-01
	9.00E-01	9.90E-01
	1.00E 00	1.00E 00

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## APPENDIX A

### STEP AND RAMP RESPONSE CAPABILITY FOR NASAP

The algorithm used by NASAP to determine the impulse response of a circuit can be easily extended to find the step and ramp response. Listed below are the modifications that are necessary to accomplish this. Note that if, for example, both the impulse and step responses are required for a particular circuit, the problem must be executed twice.

To the COMMON/FREQ/ cards of subroutines NASINP and BODE must be added:

,KRESP

The following additions must be included in subroutine NASINP.

After card 1010, three cards must be added:

Go to 3  
824 WRITE (6, 117)  
117 FORMAT (35H INCLUDE TYPE OF TIME RESPONSE CARD)

1011  
1012  
1013

After card 1180, nine cards must be added:

READ (5,101) INLST	1181	
WRITE (6,107) INLST	1182	
KRESP = 1000	1183	
IF (ICHAR(INLST(1)). EQ. 49. AND ICHAR(INLST(2))EQ.54)	KRESP = 1	1184
IF (ICHAR(INLST(1)). EQ. 55. AND ICHAR(INLST(2)).EQ.56)	KRESP = 1	1185
IF (ICHAR(INLST(1)). EQ. 62)	KRESP = 2	1186
IF (ICHAR(INLST(1)). EQ. 59 AND ICHAR(INLST(2)).EQ.41	KRESP = 3	1187
IF (KRESP. EQ.1000) GO TO 824	1189	
WRITE (6,116)	1189A	

The following changes must be made in INV:

Change card 14930  
From DIMENSION F(100), T(100), COE(2)  
To DIMENSION F(100), T(100), COE(2), TYPE (2,3)  
Change card 14990  
From WRITE (6,50)  
To WRITE (6,50) (TYPE(J,KRESP), J=1,2)  
Change card 1500  
From 50 FORMAT (26H IMPULSE RESPONSE FUNCTION //7H F(t)=)  
To 50 FORMAT (1H, 2A4, 18H RESPONSE FUNCTION //7H F(t)=)  
Change card 15280  
From WRITE (6,40) TRNS,(T(K), F(K), K=1, 51)  
To WRITE (6,40) (TYPE(J,KRESP), J=1,2), TRNS, (T(K), F(K), K=1, 51)  
Change card 15290  
From 40 FORMAT (17H IMPULSE RESPONSE // 5H TIME, 18X, 10A1/(E12,4,E23.8))  
To 40 FORMAT(1H,2A4, 9H RESPONSE // 5H TIME, 18X, 10A1/(E12,4,E23.8))  
Change card 15300  
From WRITE (6,41)

cont'd.

```
To WRITE (6,41) (TYPE(J,KRESP), J=1,2)
Change card 15310
From 41 FORMAT (1H1, 45X, 17HIMPULSE RESPONSE, /)
To 41 FORMAT (1H1, 45X, 2A4, 9H RESPONSE, /)
```

The following additions must be included in subroutine AINV.

After card 14930, two cards must be added:

DATA TYPE /4HIMPU,4HLSE, 4HSTEP, 4H ,4HRAMP, 4H	14931
COMMON/FREQ/THI, FEQ (3), CC,Q, KRESP	

After card 15060, twelve cards must be added:

76 Go To (77,76,78,76), KRESP	15061
IAD2=IAD2+1	15062
RootR (IAD2,2) = 0	15063
RootI (IAD2, 2) = 0	15064
78 Go To 77	15066
IAD2=IAD2+1	15067
RootR(IAD2,2) = 0	15068
RootI(IAD2,2) = 0.001	15069
IAD2= IAD2+1	15069A
RootR (IAD2,2) = 0	15069B
RootI (IAD2,2) = -0.001	15079C
77 CONTINUE	15069E

When the above modifications are incorporated in the NASAP program package, the only change from a user's point of view is that he must include one card, immediately following the NASAP PROBLEM card, on which is punched (beginning in column one) one of the four following comments:

```
NO RESPONSE
IMPULSE RESPONSE
STEP RESPONSE
RAMP RESPONSE
```

The choice of which comment is to be used depends on what type of time response, if any, is desired. In actuality, only the first two letters of each of the above comments are really necessary. In cards 1181 through 1189A of subroutine NASINP, the program reads the card after the NASAP PROBLEM card and then prints it. If the letters in columns one and two are on R and a A respectively, the time response variable KRESP is set equal to 3. If there is an N in column one and an O in column two, KRESP then is set equal to 1. If there is an I in column 1 and an M in column two KRESP is set equal to 1. If there is an S in column one, KRESP then has a value of 2. If the program encounters none of the above letter combinations on the second input card, it will print out the error message given on card 1013 and about the job.

The variable KRESP is used in subroutine INV to select the proper elements of the two-dimensional matrix TYPE such that the headings of the printout of the time response agree with the information on the second input card. Thus, if a step response is desired(KRESP=2), the elements in the second column of TYPE will be printed. The variable KRESP is also used in INV to select the proper poles that must be included with the transfer function poles to yield the particular time response. On lines 15061, if KRESP=1(impulse response), the

## APPENDIX B

Suggested Revision that combines Subroutines SENS & SENSS  
of NASAP 69/I into a single shorter Subroutine SENSS

In the University of Pennsylvania monthly report for July 1969,  
the possibility was mentioned of eliminating subroutine SENS by using  
the results of the calculations of subroutine SENSS (namely  $S_X^H$ ) and the  
four equations.

$$S_X^{|H|} = \operatorname{Re} S_X^H$$

$$S_X^{\phi} = \frac{1}{\phi} \cdot \operatorname{Im} S_X^H$$

$$S_X^{\operatorname{Re} H} = \operatorname{Re} S_X^H - \left( \frac{\operatorname{Im} H}{\operatorname{Re} H} \cdot \operatorname{Im} S_X^H \right)$$

$$S_X^{\operatorname{Im} H} = \operatorname{Re} S_X^H + \left( \frac{\operatorname{Re} H}{\operatorname{Im} H} \cdot \operatorname{Im} S_X^H \right)$$

where  $S_X^H = \operatorname{Re} S_X^H + j \operatorname{Im} S_X^H$

and  $H(jw) = |H| e^{j\phi} = \operatorname{Re} H + j \operatorname{Im} H$ .

This can be accomplished by making the following additions to  
subroutine SENSS. After card 17000, add the following 3 cards:  
cards 15370, 15380, and 15390 from SENS. After card 17030, remove  
cards 17040, 17050, and 17060 and add the following 6 cards:

COMMON/LOOPS/DM(100), PHIH(250), LOGF(250), LGBSNS(250), ABSH(250),  
1 ABSENS(250), REH(250), IMH(250), PHISNS(250), CSENS(250), SENSRE(250),  
card 15440 from SENS  
card 15450 from SENS with DUM1(288) changed to DUM1(38)

card 15490 from SENS

card 15500 from SENS with the addition of ,LGBSNS.

After card 17110, add the following 2 cards: cards 15520 and 15530 from SENS.

After card 17540, add the following 39 cards:

card 15540 from SENS

card 15550 from SENS with 1HO changed to 1HL.

card 15560 from SENS

Do 802 I=1, NITR

Z = 10.\*\* LOGF(I)

RE = REAL (CSENS(I))

AIM = AIMAG (CSENS(I))

CALL QZERO (REH,I,SENSRE,I,GLAG, \$81, \$82)

81 SENSRE (I) = RE - (IMH(I)/REH(I)) \* AIM

82 CALL QZERO (IMH,I,SENSIM,I,GLAG, \$83, \$84)

83 SENSIM (I) = RE + (REH(I)/IMH(I)) \* AIM

84 SENBS (I) = RE

97 CALL QZERO (PHIH,I,SENSFI,I,GLAG, \$98, \$99)

98 SENSFI (I) = (180./PI) \* (AIM/PHIH(I))

99 WRITE (6,180) LOGF(I),Z,SENSRE(I),SENSIM(I),SENBS(I),SENSFI(I)

180 FORMAT (5X,6(E 15.7, 2X))

802 CONTINUE

cards 16110 thru 16320 from SENS (22 cards)

The following 10 cards of SENSS must be slightly modified:  
(NOTE - etc. means that the rest of the card remains unchanged).

Change card 17120

from WRITE(6,100)

to WRITE(6,200)

card 17130

from 100 FORMAT(1HL, etc.

to 200 FORMAT(1HO, etc.

card 17390

from 96 WRITE(6,105) etc.

to 96 WRITE(6,205) etc.

card 17400

from 105 FORMAT etc.

to 205 FORMAT etc.

card 17450

from WRITE(6,102)

to WRITE(6,202)

card 17460

from 102 FORMAT etc.

to 202 FORMAT etc.

card 17470

from WRITE(6,103) etc.

to WRITE(6,203) etc.

card 17480

from 103 FORMAT etc.  
to 203 FORMAT etc.

card 17520

from WRITE(6,102)  
to WRITE(6,202)

card 17530

from WRITE(6,103) etc.  
to WRITE(6,203) etc.

Finally in BODE remove card 13510. In SPLIT the following 2 minor modifications are needed:

On card 16670

change ARR(1000), DUMM2(1750)  
to ARR(750) , DUMM2(2250)

On card 16690

change DUM1(288)  
to DUM1( 38)

These changes essentially delete cards 15570 thru 16100 from SENS.

The calculations in this block of cards are replaced by the Do-Loop (labelled 802) given above. The remaining cards of SENS are then inserted into subroutine SENSS in the appropriate locations. On the RCA Spectra 70, the original version of SENS requires 4000 bytes of memory and the original version of SENSS requires 2540 bytes. However the suggested new version of SENSS requires 4392 bytes - thus realizing a saving of 2148

bytes. Enclosed is a listing of the new version of SENSS. Due to the use of a BCD coded program on a EBCDIC machine, the following symbols are equivalent:

#	is equivalent to	=		
%	"	"	"	(
<	"	"	"	)
&	.	"	"	+

Note that the printed output of the original SENSS now precedes, the printed output of the original version of SENS.