```
NASA TECHNICAL
NASA TM X-62,248 miEMORANDLM
```



THE TURBULENT MEAN-FLOW, REYNOLDSSTRESS. AND HEAT-FLUX EQUATIONS IN MASS-AVERAGED DEPENDENT VARIABLES

Morris W. Rubesin and William C. Rose

## Ames Research Center Moffett Field Calif. 94035

## TABLE OF CONTENTS

Page No.
INTRODUCTION ..... 1
SYMBOLS ..... 4
TURBULENLE EQUATIONS IN ORTHOGONAL TENSOR FORM ..... 5
Mean 'low Equations ..... 8
Iurbulence Growth Equatiors ..... 0
IHE AXISYMETRIC-FLOW EQUATIONS ..... 11
Mean Flow Equations ..... 11
Turbulence crowth Equations ..... 14
COMMENTS ON THE USE OF MASS-AVERAGED VARIABLES ..... 18
APPENDIX A. Orthogonal Tensor Relationships ..... 20
Axisymmetric Case ..... 22
APPENDIX B. The Relationships Between the Reynolds and the Mass- ..... 24
Averaged Dependent Variabies


#### Abstract

This note presents the time-dependent, turbulent mean-flow, Reynolds stress, and heat flux equations in mass-averaged dependent variables. These equations are given in conservative form for both seneralized orthogonal and axisymetric coordinates. For the case of small viscosity and thermal con:ductivity fluctuations, these equations are considerably simpler than the general Reviolds system if dependent variables for - compressible fluid and permit a more direct extension of low speed turbulence modeling to computer codes describing high speed turbulence fieids.


## INTRODUCTION

In recent years, significant advancements in both computation techniques and computer hardware have led to sophisticated prediction techniques for describing complex compressible fluid flows. The most general techniques avoid boundary-layer approximations and solve the complete time-dependent, conservation equations of mass, momentum and thermal energy, and possibly species. (See ref (1) and (2).) The transport mechanisms that have been employed in these techniques are consistent with melecular ol laminar processes. Methods such as those presented by MacCormack or Carter have been shown to apply to a flow as complex as a shock-wave boundary-layer interaction, even when the adverse pressure gradient generated by the shock wave is strong enough to cause separation of the boundary-lajer flow.

The sciccessful methods for laminar flows have not, however, been carried over to the computation of turbulent flows - particularly strongly interacting flows. Turbulent-flow solutions, which are presumably contained in the instantaneous viscous flow conservation equations, cannot be obtained at present because the small grid size required for accurately resolving the decay processes of turbulence cannot be accommodated in present computers. Therefore, these instantaneous equations must be time averaged to obtain a system of equations with spatial characteristics consistent with present computers. In the time averaging process there is a loss of phase and wave-length information of the turbulent motions. Although this information may be critical to the ultimate understanding of turbulence, time averaging has yielded most of the significant results related to the mean flow in boundary layers and may prove equally valuable for the more complex flow fields of current interest.

The first step i establishing time-iveraged turbulent flow equations is the expansion of all the dependent variables into the sum of two terms, one slowly varying and the other randomly varying at the high frequencies characteristic of turbulence. With compressible flows there is some freedom in the choice of how this expansion is accomplished. For example, instantaneous velocity components san be expanded either as the sum of two velocities. as in the classical Reynolds expansion, or as the sun of two mass fluxes divided by the mean density, the mass averaged dependent variable expansion, first suggested by Favre (ref. 3). As an example of Reynolds' expansion, Donaldson (ref. 4) has written the complete system of equations for compressible flow in generalized tensor form so tha: they can be particularized to any desired coordinate system. Donaldson's equations explicitly contain density fluctuations as well as velocity fluctuations, thus, adding considerable complexity as compared with the constant property case. This added complexity can be reduced, as shown in ref. 5 and 6, if the expansion of the dependent variables is carried out using mass averaged exparsions. These equations essentially exhibit a term by term correspondence with the constant property case. This correspondence can be important if the basic turbulence mechanisms in compressible flows remain similar to those in constant property flows (the so-called Morkovin hypothesis), thereby permitting convenient extension of the large body of experience existing with incompressible fiows.

The work of refs. 5 and 6 is limited, however, in that the Reynolds stress and turbulent heat flux equations were not presented and, further, the equations were confined to the steady state and expressed in cartesian coordinates. The
general objective of this note is to remove these limitations by

1. adding the iime-dependent terms to all the equations
2. presenting the Reynolds stress and heat flun equations
3. employing a generalized orthogonal coordinate system
4. casting the equations in conservative form, cunsistent with existing computer codes (refs. 1 and 2)
and
5. particularizing the equations to cylindrical coordinates for comparison with axisymmetric fiow data (e.g. ref. 7).
e total internal energy per unit of volume, $\rho h-p+\rho \frac{u_{j} u_{j}}{2}$
h specific enthalpy
H total specific enthalpy, $h+\frac{u_{j} u_{j}}{2}$
p static pressure
P period of integration for turblence quantities
Pr molecular Prandtl number
$q_{j}$ heat flux in $j$ th direction, $-\lambda \frac{\partial T}{\partial x_{j}}$ or $-\frac{\mu}{\operatorname{Pr}} \frac{\partial h}{\partial x_{i}}$
s dumny variable of integration
$t$ time
T temperature
$u_{i} \quad$ velocity component in $i$ th direction
$\lambda$ molecular thermal conductivity
$\mu$ molecular viscosity
p density
$\tau_{i j}$ ins*antaneous shear stress in $i$ th direction on a surface normal to the $j$ th direction, $\mu\left[\left(\frac{\partial u_{i}}{\partial x_{j}}+\frac{\partial u_{j}}{\partial x_{i}}\right)-\frac{2}{3} \delta_{i j} \frac{\partial u_{k}}{\partial x_{k}}\right]$

TURBULENCE EQUATIONS IN ORTHOGONAL TENSOR FORM
In the derivation of the mean flow and turbulence equations, orthogonal tensor notation will be employed. The tui conventions of a repeated subscript indicating summation over the entire range of the indices and a comma representing partial differentiation will be used to make the equations compact. Orthogonal tensors were adopted to permit working directly with the physical components, thereby avoiding the need for identifying specific tensors as being covariant or contravariant. The use of orthogonal coordinate systems does not cause significant loss of generality in most fluid mechanical applications.

The time-dependent, conservation equations for a compressible fluid with zero external body forces are

Continuity

$$
\begin{equation*}
\rho, t+\left(\rho u_{j}\right), j=0 \tag{1}
\end{equation*}
$$

Momentum in the eth direction ( $i=1,2$, or 3 )

$$
\begin{equation*}
\left(\rho u_{i}\right)_{, t}+\left(\rho u_{i} u_{j}+\delta_{i j} p-\tau_{i j}\right), j=0 \tag{2}
\end{equation*}
$$

Thermal Energy

$$
\begin{equation*}
(\rho H-p)_{, t}+\left(\rho u_{j} H+q_{j}-u_{i} \tau_{i j}\right), j=0 \tag{3}
\end{equation*}
$$

to convert these equations to time averaged equations, the following expansions of the dependent variables are employed:

$$
\begin{align*}
\rho & =\bar{p}+\rho^{\prime} \\
p & =\bar{p}+p^{\prime} \\
\tau_{i j} & =\bar{\tau}_{i j}+\tau_{i j}^{\prime}  \tag{4}\\
q_{j} & =\bar{q}_{j}+q_{j}^{\prime}
\end{align*}
$$

-5-
and the mass averaged dependent variables

$$
\begin{align*}
& u_{i}=\tilde{u}_{i}+u_{i}^{\prime \prime} \\
& H=\tilde{H}+H^{\prime \prime}  \tag{5}\\
& h=\tilde{h}+h^{\prime \prime}
\end{align*}
$$

The superscripts - , represent time averaging over a period $P$ that is long compared to the fluctuation period of the turbulence but short compared to the time constant for changes within the flow field as a whole. The superscript ' or " represents the instantaneous fluctuation about the corresponding time averaged quantity. Thus, we set

$$
\begin{equation*}
\bar{A}\left(t_{0}\right)=\frac{1}{P} \int_{t_{0}-(P / 2)}^{t_{0}+(P / 2)} A(t) d t \tag{6}
\end{equation*}
$$

then,

$$
\begin{equation*}
\frac{\overline{\partial A\left(t_{0}\right)}}{\partial t}=\frac{1}{P} \int_{t_{0}-(P / 2)}^{t_{0}+(P / 2)} \frac{\partial A(t)}{\partial t} d t=\frac{\partial}{\partial t} \overline{A\left(t_{0}\right)} \tag{7}
\end{equation*}
$$

For example, from Eqs. (4) and (6)

$$
\begin{align*}
\bar{\rho}\left(t_{0}\right) & =\frac{1}{P} \int_{t_{0}-(P / 2)}^{t_{0}+(P / 2)}\left(\bar{\rho}\left(t_{0}\right)+\rho^{\prime}(t)\right) d t  \tag{8}\\
& =\bar{\rho}\left(t_{0}\right)+\bar{\rho}^{\prime}\left(t_{0}\right) \tag{9}
\end{align*}
$$

From Eq. (9),

$$
\begin{equation*}
\bar{p}^{\prime}=0 \tag{10}
\end{equation*}
$$

Thu's, in general

$$
\begin{equation*}
\bar{A}^{\prime}=\frac{1}{P} \int_{t_{0}-(P / 2)}^{t_{0}+(P / 2)} A^{\prime}(t) d t=0 \tag{11}
\end{equation*}
$$

a single fluctuating quantity raised to the first power averages to zero.
The mass-averaged dependent variables are defined (see Appendix B) such that the mean velocity is

$$
\begin{equation*}
\tilde{u}_{i}=\bar{u}_{i}=\frac{\overline{\rho u_{i}}}{\bar{p}} \tag{12}
\end{equation*}
$$

and the fluctuating velocity averages to

$$
\begin{equation*}
\overline{\rho u_{i}^{\prime}}=0 \tag{13}
\end{equation*}
$$

From Eqs. (4), (5) and (13)

$$
\begin{equation*}
u_{i}^{\bar{\prime}}=-\frac{\overline{\rho^{\prime} u_{i}^{\prime \prime}}}{\bar{\rho}} \tag{14}
\end{equation*}
$$

so that $\overline{u_{i}^{\prime \prime}}$ is not zero in contrast to Eq. (11).
Similarly, the mass averaged static enthalpy is

$$
\begin{equation*}
\tilde{h}=\frac{\overline{\rho h}}{\bar{\rho}} \tag{15}
\end{equation*}
$$

and

$$
\begin{equation*}
\overline{\rho h^{\prime \prime}}=0 \tag{16}
\end{equation*}
$$

Again $\overline{h^{"}}$ is not zero, but rather

$$
\overline{h^{\prime \prime}}=-\frac{\overline{\rho^{\prime} h^{\prime \prime}}}{\bar{\rho}}
$$

## Mean Flow Equations

When the instantaneous density and velocity, express as Eys. '4) and (5), are inserted into the continuity equation (1) and the resuiting equation is time averaged according to Eq. (6), there results

$$
\begin{equation*}
\bar{\rho}_{, t}+\left(\bar{\rho}_{i}\right)_{, j}=0 \tag{17}
\end{equation*}
$$

This time-averaged continuity equation has identical form to Eq. (1) except that the instantaneous quantities $\rho$ and $u$ are replaced by the time averaged quantities $\vec{\rho}$ an .

Similar operations of expanding the dependent variables arcording to Eqs. (4) and (5) and then Reynolds averaging converts the momentum equations, Eq. (2), and the energy equation, Eq. (3), to

$$
\begin{equation*}
\left(\bar{\rho}_{i}\right)_{, t}+\left[\bar{\rho}_{j} \tilde{u}_{i}+\delta_{i j} \bar{p}-\left(\bar{\tau}_{i j}-\overline{\rho u_{i}^{\prime \prime \prime}}{ }^{\prime \prime}, j\right], j=0\right. \tag{18}
\end{equation*}
$$

and
$(\bar{\rho} \tilde{H}-\bar{p}), t+\left[\bar{\rho} \tilde{u}_{j} \tilde{H}+\bar{q}_{j}+\overline{\rho u_{j}^{\prime \prime} h^{\prime \prime}}-\tilde{u}_{i}\left(\bar{\tau}_{i j}-\overline{\rho u_{i}^{\prime \prime} u_{j}^{\prime \prime}}\right)-u_{i}^{\prime \prime}\left(\tau_{i j}-\frac{\rho u_{i}^{\prime \prime} u_{j}^{\prime \prime}}{2}\right)\right], j=0$

Again, their form is generally identical with their laminar fiow counterparts excert for the addition of the Reynolds stress, heat flux, and the additional mean energy dissipation terms given below:

$$
\begin{equation*}
\text { Reynolds stress }=\overline{\tau_{i j}^{t}}=-\overline{\rho v_{i}^{\prime} u_{j}^{\prime \prime}} \tag{20}
\end{equation*}
$$

$$
\begin{align*}
\text { Reynolds heat flux } & =\overline{q_{j}^{t}}=\overline{\rho u_{j}^{\prime \prime} h^{\prime \prime}}  \tag{21}\\
\text { Mean energy dissipation } & =\overline{u_{i}^{\prime \prime}\left(\tau_{i j}-\frac{\rho u_{i}^{\prime \prime u_{j}^{\prime \prime}}}{2}\right)} \tag{22}
\end{align*}
$$

Turbulence Growth Equations
The growth of the Reynolds stresses are found by manipulation of Eqs. (1)
and (2) into a moment of momentism equation

$$
\left(\rho u_{i} u_{k}\right), t+\left(\rho u_{i} u_{k} u_{j}\right), j+u_{k} p_{, i}+u_{i} p_{, k}-u_{k} \tau_{i j, j}-u_{i} \tau_{k j, j}=0
$$

This equation is ther expanded according to Eqs. (4) and (5) and time averaged according to Eq. (6) to yield the sum of the mear flow terms

$$
\begin{align*}
& \left(\bar{\rho} \tilde{u}_{i} a_{k}\right), t+\left(\bar{\rho} \tilde{u}_{j} a_{i} a_{k}\right), j \\
& =-\tilde{a}_{k} \bar{p}_{, i}-{a_{i}}_{i} \bar{p}_{, k}+\tilde{a}_{k}\left(\bar{\tau}_{i j}-\overline{\rho u_{i}^{\prime \prime} u_{j}^{\prime \prime}}\right)_{, j}+{\tilde{u_{i}}}_{i}\left(\bar{\tau}_{k j}-\overline{\rho u_{k}^{\prime \prime} u_{j}^{\prime \prime}}\right), j \tag{24}
\end{align*}
$$

ani the turbulence quaniities (growth of Reynolds stresses)

$$
\begin{align*}
\left(\overline{\rho u_{i}^{\prime \prime} u_{k}^{\prime \prime}}\right)_{, t}+\left(\tilde{u}_{j} \overline{\rho u_{i}^{\prime \prime} u_{k}^{\prime \prime}}\right), j= & -\left(\overline{\rho u_{i}^{\prime \prime} u_{j}^{\prime \prime}}\right) u_{k, j}-\left(\overline{\rho u_{k}^{\prime \prime} u_{j}^{\prime \prime}}\right) u_{i, j}-\left(\overline{\rho u_{i}^{\prime \prime} u_{k}^{\prime \prime} u_{j}^{\prime \prime}}\right), j \\
& -\left(\overline{u_{k}^{\prime \prime \prime} p}\right)_{, i}-\left(\overline{u_{i}^{\prime \prime p}}\right)_{, k}+\overline{p\left(u_{k, i}^{\prime \prime}+u_{i, k}^{\prime \prime}\right)} \\
& +\left(\overline{u_{k}^{\prime \prime} \tau}\right)_{, j}+\left(\overline{u_{i}^{\prime \prime} \tau_{k j}}\right)_{, j}-\overline{\tau_{i j}} \cdot{ }^{1}-\overline{\tau_{k j}} \bar{u}_{i, j}^{\prime \prime \prime} \tag{25}
\end{align*}
$$

Eq. (24) follows directly from Eqs. (17) and (18); so that the separation of Eq. (24) and Eq. (25) as indicated is not arbitrary. Eq. (25) corresponds term by term to the constant property equation in Ref. 8, except with $\rho$ occurring inside of the operator signs of time-averaging and differentiation.

The turbulence kinetic energy equation, with

$$
\begin{equation*}
k=\frac{1}{2 \bar{\phi}} \rho u_{i}^{\prime \prime} u_{i}^{\prime \prime} \tag{26}
\end{equation*}
$$

and

$$
\begin{equation*}
\bar{k}=\frac{1}{2 \bar{\rho}}\left(\overline{o u_{i}^{\prime \prime} u_{i}^{\prime \prime}}\right) \tag{27}
\end{equation*}
$$

is
$(\overline{\rho \bar{k}})_{, t}+\left(\bar{\rho} \tilde{u}_{j} \ddot{k}\right), j=-\left(\overline{\rho u_{i}^{\prime \prime} u_{j}^{\prime \prime}}\right) \tilde{u}_{i, j}-\left(\overline{\rho u_{j}^{\prime \prime k}}\right)_{, j}-\left(\overline{u_{i}^{\prime \prime p}}\right)_{, i}+\overline{\rho u_{i, i}^{\prime \prime}}+\left(\overline{u_{i}^{\prime \prime} \tau}{ }_{i j}\right), j-\overline{\tau_{i j} u_{i, j}^{\prime \prime}}$

The terms on the right are, sequentially, the turbulence production, the turbulent diffusion, the pressure diffusion, pressure work, molecular diffusion, and dissipation of the turbulence kinetic energy. Eq. (28) is consistent wath Rotta's and Townsend's (Ref. 9) constant property equations.

The equation for the growth of Reynolds heat flux, Eq. (21), follows from the energy equation, Eq. (3), combined with Eq. (23) with $i=k$ and the equation resulting from $h$ times Eq. (?)

$$
\begin{align*}
& \left(\rho u_{i} h\right), t+\left(\rho u_{i} u_{j} h ;, j-u_{i} p_{, t}-u_{i} u_{k} p_{, k}+h p_{, i}\right. \\
& \tag{29}
\end{align*}
$$

Again, when this equation is expanded according to Eqs. (4) and (5), and the resulting equation is time averaged according to Eq. (6), there is obtained the sum of the mean quantities

$$
\begin{aligned}
& \left(\bar{\rho} \bar{h} \tilde{u}_{i}\right), t+\left(\bar{\rho} \bar{u}_{j} \tilde{u}_{i} \bar{h}\right), j-\bar{u}_{i} \bar{p}_{, t}+\bar{u}_{i}\left[\bar{q}_{j}+\overline{\rho v_{j}^{\prime \prime} h^{\prime \prime}}-\bar{u}_{k} \bar{\tau}_{k j}-\overline{u_{k}^{\prime \prime} \tau_{k j}}\right], j
\end{aligned}
$$

and Reynoids heat flux terms

$$
\begin{aligned}
& \left(\overline{\rho u_{i}^{\prime \prime} h^{\prime \prime}}\right), t+\left(u_{j} \overline{\rho u_{i}^{\prime \prime} h^{\prime \prime}}\right), j+\left(\overline{\rho u_{j} h^{\prime \prime}}\right) \tilde{u}_{i, j}+\left(\overline{\rho u_{i}^{\prime \prime} u_{j}^{\prime \prime}}\right) \tilde{h}_{, j}+\left(\overline{\rho u_{j}^{\prime \prime} u_{i}^{\prime \prime} h^{\prime \prime}}\right), j-\overline{u_{i}^{\prime \prime}}, t \\
& -\overline{u_{k}} \overline{u_{i}^{\prime \prime} F_{, k}}-\overline{u_{i}^{\prime \prime} u_{i}^{\prime \prime} p, k}+\overline{h^{\prime \prime} p}, \overline{h^{\prime \prime} \tau_{i j, j}}+\overline{u_{i}^{\prime \prime} q_{j, j}}-\bar{u}_{k, j} \overline{\bar{u}_{i}^{\prime \prime} \tau_{k i}}-\overline{u_{k, j}^{\prime \prime} u_{i}^{\prime \prime} \tau_{k j}}=0
\end{aligned}
$$

The proper separation of Eqs. (30) and (31) follows fror the use of moments of Eqs. (18) and (19) to yield īq. (30).

## THE AXISYMETRIC-FLOW EQUATIONS

The equations given above in orthogonal tensor form are particularized in this section to an axisymetric system (i.e. a cylindrical coordinate system with $\frac{\partial(\overline{)}}{\partial \theta}=0$ and $\tilde{w}=0$ ). These equations are obtained using the particularized expansions and coordinate nomenclature given in the Appendix.

## Mean Flow Equations

The mean flow equations Eq. (17) through (19) particularize to Continuity:

$$
\begin{equation*}
\frac{\partial(r \bar{\rho})}{\partial t}+\frac{\partial(r \bar{\rho} \bar{u})}{\partial x}+\frac{\partial(r \vec{\rho} v)}{\partial r}=0 \tag{32}
\end{equation*}
$$

## x-momentum

$$
\begin{equation*}
\frac{\partial}{\partial t}(r \bar{\rho} \bar{u})+\frac{\partial}{\partial x}\left[r\left(\bar{\rho} \bar{u} u \bar{u}+\bar{p}-\bar{\tau}_{x x}+\overline{\rho u^{\prime \prime} u^{\prime \prime}}\right)\right]+\frac{\partial}{\partial r}\left[r\left(\bar{\rho} \overline{v u}-\bar{\tau}_{x r}+\overline{\rho u^{\prime \prime} v^{\prime \prime}}\right)\right]=0 \tag{33}
\end{equation*}
$$

## r-momentum

$$
\begin{align*}
& \frac{\partial}{\partial t}(r \bar{\rho} \tilde{v})+\frac{\partial}{\partial x}\left[r\left(\bar{\rho} \bar{u} v-\overline{\tau_{r x}}+\overline{\rho u^{\prime \prime} v^{\prime \prime}}\right)\right]+\frac{\partial}{\partial r}\left[r\left(\bar{\rho} \bar{v} \bar{v}+\bar{p}-\overline{\tau_{r r}}+\overline{\rho v^{\prime \prime} v^{\prime \prime}}\right)\right] \\
&+\overline{\tau_{\theta \theta}}-\bar{p}-\overline{\rho w^{\prime \prime} w^{\prime \prime}}=0 \tag{34}
\end{align*}
$$

Total thermal energy

$$
\begin{align*}
& \frac{\partial}{\partial t}(r \bar{e})+\frac{\partial}{\partial x} r\left[\tilde{u}(\bar{e}+\bar{p})+\bar{q}_{x}+\overline{\rho u^{\prime \prime \prime} h^{\prime \prime}}-\bar{u}\left(\bar{\tau}_{x x}-\overline{\rho u^{\prime \prime} u^{\prime \prime}}\right)-v\left(\bar{\tau}_{r x}-\overline{\rho v^{\prime \prime} u^{\prime \prime}}\right)-\overline{u^{\prime \prime}\left(\tau_{x x}-\frac{\rho u^{\prime \prime} u^{\prime \prime}}{2}\right.}\right) \\
& \left.-\overline{v^{\prime \prime}\left(\tau_{r X}-\frac{\rho v^{\prime \prime} u^{\prime \prime}}{2}\right.}\right)-\overline{\left.w^{\prime \prime}\left(\tau_{\theta x}-\frac{\rho w^{\prime \prime} u^{\prime \prime}}{2}\right)\right]+\frac{\partial}{\partial r} r\left[\tilde{v}(\tilde{e}+\bar{p})+\bar{q}_{r}+\overline{\rho v^{\prime \prime} h^{\prime \prime}}, ~\right.} \\
& -\tilde{u}\left(\bar{\tau}_{x r}-\overline{\rho u^{\prime \prime} v^{\prime \prime}}\right)-\nabla\left(\bar{\tau}_{r r}-\rho v^{\prime \prime} v^{\prime \prime}\right)-\overline{\left.u^{\prime \prime}\left(\tau_{X r}-\frac{\rho u^{\prime \prime} v^{\prime \prime}}{2}\right)-\overline{v^{\prime \prime}\left(\tau_{r r}-\frac{\rho v^{\prime \prime} v^{\prime \prime}}{2}\right.}\right)} \\
& \left.-\overline{w^{\prime \prime}\left(\tau_{\theta \mathbf{r}}-\frac{\rho w^{\prime \prime} v^{\prime \prime}}{2}\right)}\right]=0 \tag{35}
\end{align*}
$$

These equetions may be written in a condensed vector form convenient for numerical computations (ref. 1 or 2) as

$$
\begin{equation*}
\frac{\partial r \vec{U}}{\partial t}+\frac{\partial r \vec{F}}{3 \vec{r}}+\frac{\partial r \vec{G}}{\partial r}+\vec{A}=0 \tag{36}
\end{equation*}
$$

where, for example,

$$
\vec{U}=\left(\begin{array}{c}
\bar{\rho}  \tag{37}\\
\bar{\rho} \bar{\alpha} \\
\bar{\rho} \bar{\nu} \\
\bar{e}
\end{array}\right) \text { and } \vec{A}=\left(\begin{array}{c}
0 \\
0 \\
\bar{\tau}_{\theta \theta}-\bar{p}-\overline{\rho w^{\prime \prime} w^{\prime \prime}} \\
0
\end{array}\right)
$$

These equations may also be written in an alternative form by dividing through by r to give

$$
\begin{equation*}
\frac{\partial \vec{U}}{\partial t}+\frac{\partial \vec{F}}{\partial x}+\frac{\partial \vec{G}}{\partial r}+\frac{1}{r}(\vec{G}+\vec{A})=0 \tag{38}
\end{equation*}
$$

We may combine the two inhomogeneous terms as

$$
\vec{B}=\frac{\vec{G}+\vec{A}}{\mathbf{r}}
$$

In this form $\vec{B}$ becomes the following vector:


In the alternative form of the equations, the transformation from the axisymmetric case to the planar flow case is simple in that the vector $B$ vanishes and the remainder of the equation is unchanged, i.e., we have for the planar case

$$
\begin{equation*}
\frac{\partial \vec{U}}{\partial t}+\frac{\partial \vec{F}}{\partial x}+\frac{\partial \overleftrightarrow{G}}{\partial r}=0 \tag{40}
\end{equation*}
$$

## Turbulence Growth Equations

To affect a closure of the above equations (in either of the two forms presented) we require equations for $\overline{\rho u^{\prime \prime 2}}, \overline{\rho u^{\prime \prime} v^{\prime \prime}}, \overline{\rho v^{\prime \prime 2}}, \overline{\rho w^{\prime \prime 2}}, \overline{\rho u^{\prime \prime} h^{\prime \prime}}, \overline{\rho v^{\prime \prime} h^{\prime \prime}}$ as well as tine terms in the thermal energy equation which have the form:

$$
\overline{u_{i}^{\prime \prime}\left(\tau_{i j}-\frac{1}{2} \rho u_{i}^{\prime \prime} u_{j}^{\prime \prime}\right)}
$$

Relative to these latter terms, Laufer (ref. 5) points out that, although all previous analyses have neglected them, they can be of the same order as the turbulence production terms.
Th e Reynolds stress equations in the axisymmetric form are



$+\delta_{i 3}\left(\overline{\rho u_{k}^{i \prime} v^{\prime \prime} w^{i \prime}}-\overline{\tau_{r \theta^{\prime}} u_{k}^{i \prime}}\right)-\delta_{i 2}\left(\overline{\rho u_{k}^{i n} w^{\prime \prime} w^{i \prime}}-\overline{\tau_{\theta \theta}} \bar{u}_{k}^{i \prime}+\overline{u_{k}^{i \prime} p}\right)$
$\delta _ { k } \longdiv { \rho u _ { i } ^ { 1 / } w ^ { 1 t } }$
(41)
$\left.\begin{array}{l}\delta_{i 3} \overline{\rho u_{k}^{\prime \prime} w^{\prime \prime}} \hat{v} \\ +\overline{\tau_{k \theta}} \frac{\partial u_{i}^{\prime \prime}}{\partial \theta}\end{array}\right]$


$\left.{ }^{00_{k}}\right]$

$\tau_{k r} \frac{\partial u_{i}^{\prime \prime}}{\partial r}$
$\frac{\partial u_{k}}{\partial x}+\overline{\rho u_{i}^{\prime!} v^{\prime \prime}}$

ami

kS r $\square$

The equation for the turbulence kinetic energy in axisymmetric form is


$$
+\mathbf{x}\left[\overline{\mathbf{p} \frac{\partial u^{i \prime}}{\partial x}}+\overline{p \frac{\partial v^{\prime \prime}}{\partial r}}+\overline{\bar{p} \frac{\partial w^{\prime \prime}}{\partial \theta}}+\overline{\overline{v^{\prime \prime} p}} \overline{\mathbf{r}}\right]
$$

The equations for the rurbulent heat flux in axisymmetric form are

where $i$ takes on values 1 and 2 , for heat flux in axial and radial directions, respectively.

## COMMENTS ON THE USE OF MASS AVERAGED VARIABLES

The use of the mass-averaged variables results in mean flow turbulence equations that contain a minimum number of terms, since terms involving $\rho^{\prime}$ do not appear explicitly, and posses!, almost a term by term correspondence with their incompressible counterparts. On the other hand, the complete expressions for the molecular shear terms $\tau_{i j}$ :ontain extra terms. It will be shown that this latter condition does not ser; ously complicate natters.

Instantaneously,

$$
\begin{equation*}
\tau_{i j}=\mu\left[\left(u_{i, j}+u_{j, i}\right)-\frac{2}{3} \delta_{i j} u_{k, k}\right] \tag{44}
\end{equation*}
$$

when the Stokes criterion for the second visccsity coefficient is invoked. For compactness, we write the quantity in the brases as $S_{i j}$ so that

$$
\begin{equation*}
\tau_{i j}=\mu S_{i j} \tag{45}
\end{equation*}
$$

When Eq. (45) is expanded in the Favre variables, there results

$$
\begin{equation*}
\tau_{i, j}=\bar{\tau}_{i j}+\tau_{i j}^{\prime}=\left(\tilde{\mu}+\mu^{\prime \prime}\right)\left(\tilde{S}_{i j}+S_{i j}^{\prime \prime}\right) \tag{46}
\end{equation*}
$$

Here the viscosity is identified with the mass averaged enthalpy of Eq. (15) or the equivalent mass averaged temperatire. (See Appendix $B$ for the interrelationship of the Favre variable with the usual Reynolds expanded quantities.) From Eq. (46)

$$
\begin{equation*}
\tau_{i j}=\ddot{\tau}_{i j}+\tau_{i j}^{\prime}=\tilde{\mu} \tilde{S}_{i j}+\tilde{\mu} S_{i j}^{\prime \prime}+\mu^{\prime \prime} \tilde{S}_{i j}+\mu^{\prime \prime} S_{i j}^{\prime \prime} \tag{47}
\end{equation*}
$$

Eq. (47), time averaged, yields

$$
\begin{equation*}
\bar{\tau}_{i j}=\tilde{\mu}_{i j}+\tilde{\mu}_{i j}^{\prime \prime}+\bar{\mu}^{\prime \prime} \tilde{S}_{i j}+\bar{\mu}^{\prime \prime} S_{i j}^{\prime \prime} \tag{48}
\end{equation*}
$$

and

$$
\begin{equation*}
\tau_{i j}^{\prime}=\tilde{S}_{i j}\left(\mu^{\prime \prime}-\overline{\mu^{\prime \prime}}\right)+\tilde{\mu}\left(S_{i j}^{\prime \prime}-\overline{S_{i j}^{\prime \prime}}\right)+\mu^{\prime \prime} S_{i j}^{\prime \prime}-\overline{\mu^{\prime \prime}} S_{i j}^{\prime \prime} \tag{49}
\end{equation*}
$$

In customary Reynolds expansions, not the mass-averaged coordinates, time averages of first order turbulent quantities vanish, which would have eliminated half the terms in Eqs. (48) and (49). Thus the mass-averaged notation appears to complicii: the expressinns involving the molecular viscosity. This additional complexity is mitigated chrough arguments of the relative magnitude of the individual termis of Eqs. (48) and (49) when equations as complex as Eqs. (41) through (43) are considered for solution.

APPENDIX A

## Orthogonal Tensor Relationships

The following definitions apply for an orthogonal curvalinear coordinate s) 5 tem (Ref. 10). The metric, or length of a line element is

$$
\begin{equation*}
\mathrm{ds}^{2}=\mathrm{h}_{1}^{2} \mathrm{~d} \mathrm{x}_{1}^{2}+\mathrm{h}_{2}^{2} \mathrm{~d} x_{2}^{2}+\mathrm{h}_{3}^{2} \mathrm{~d} x_{3}^{2} \tag{A-1}
\end{equation*}
$$

where $h_{1}, h_{2}$, and $h_{2}$ are the metric elements of the orthogonal coordinates $x_{1}, x_{2}$ and $x_{3} . *$

The siadient of a scalar, a vector quantity, is given by

$$
\begin{equation*}
\phi_{, i}=\frac{1}{h_{i}} \frac{\partial \phi}{\partial x_{i}} \tag{A-2}
\end{equation*}
$$

The gradient of a vector, a tensor quantity, is

$$
\begin{equation*}
A_{i, j}=\frac{1}{h_{j}} \frac{\partial A_{i}}{\partial x_{j}}+\delta_{i j} \sum_{k} \frac{A_{k}}{h_{i} h_{k}} \frac{\partial h_{i}}{\partial x_{k}}-\frac{A_{j}}{h_{i} h_{j}} \frac{\partial h_{j}}{\partial x_{i}} \tag{A-3}
\end{equation*}
$$

where

$$
\begin{aligned}
\delta_{\perp j}(\text { Kronecker } \delta) & =1 & & i=j \\
& =0 & & i \neq j
\end{aligned}
$$

Eq. (A-3) can be applied to define the directional derivative of a vector, namely

$$
\begin{equation*}
\sum_{j} B_{j} A_{i, j}=\sum_{j} \frac{B_{j}}{h_{j}} \frac{\partial A_{i}}{\partial x_{j}}+\sum_{j} \frac{B_{i} A_{j}}{h_{i} h_{j}} \frac{\partial h_{i}}{\partial x_{j}}-\sum_{j} \frac{B_{j} A_{j}}{h_{i} h_{j}} \frac{\partial h_{j}}{\partial x_{i}} \tag{A-4}
\end{equation*}
$$

*Note that the summation convention of a repeated index is not used in this appendix.

This disagrees with Ref. (10) in that the subscripts of $A$ and $B$ in the second term on the right are interchanged there. No significant consequence occurs, however, since in Ref. (10) both B and A ultimately represent the same quantity. Again, Eq. (A-3) can be summed to yield the divergence of a vector $A_{j}$.

$$
\begin{equation*}
\sum_{j} A_{j, j}=\frac{1}{h_{1} h_{2} h_{3}}\left[\frac{\partial}{\partial x_{1}}\left(h_{2} h_{3} A_{1}\right)+\frac{\partial}{\partial x_{2}}\left(h_{1} h_{\uparrow} A_{2}\right)+\frac{\partial}{\partial x_{3}}\left(h_{1} h_{2} A_{3}\right)\right] \tag{A-5}
\end{equation*}
$$

The divergence of a tensor can be expressed from Eqs. (A-4) and (A-5) as

$$
\begin{align*}
\sum_{j} \tau_{i j, j}= & \frac{1}{h_{i} h_{2} h_{3}}\left[\frac{\partial}{\partial x_{1}}\left(h_{2} h_{3} \tau_{i 1}\right)+\frac{\partial}{\partial x_{2}}\left(h_{1} h_{3} \tau_{i 2}\right)+\frac{\partial}{\partial x_{3}}\left(h_{1} h_{12} \tau_{i 3}\right)\right] \\
& +\tau_{1 i} \frac{1}{h_{i} h_{1}} \frac{\partial h_{i}}{\partial x_{1}}+\tau_{2 i} \frac{1}{h_{i} h_{2}} \frac{\partial h_{i}}{\partial x_{2}}+\tau_{3 i} \frac{1}{h_{i} h_{3}} \frac{\partial h_{i}}{\partial x_{3}} \\
& -\tau_{11} \frac{1}{h_{1} h_{i}} \frac{\partial h_{1}}{\partial x_{i}}-\tau_{22} \frac{1}{h_{2} h_{i}} \frac{\partial h_{2}}{\partial x_{i}}-\tau_{33} \frac{1}{h_{3} h_{i}} \frac{\partial h_{3}}{\partial x_{i}} \tag{A-6}
\end{align*}
$$

Also required in the orthogonal tensor form of the equations is a tensor directional derivative analogous to the directional derivative of a vector which can be written as

$$
\begin{equation*}
\sum_{j} B_{k j} A_{i, j}=\sum_{j}\left[\frac{B_{k j}}{h_{j}} \frac{\partial A_{i}}{\partial x_{j}}+\frac{B_{k i}}{h_{i}} \frac{A_{j}}{h_{j}} \frac{\partial h_{i}}{\partial x_{j}}-\frac{B_{k j}}{h_{i}} \frac{A_{j}}{h_{j}} \frac{\partial h_{j}}{\partial x_{i}}\right] \tag{A-7}
\end{equation*}
$$

In addition, the diveggence of a third order tensor is required. ais form may be written as

$$
\begin{align*}
\sum_{j} \vec{\Sigma}_{i j k, j}= & \sum_{j}\left[\frac{1}{h_{j}} \frac{\partial E_{i j k}}{\partial x_{j}}+\frac{E_{i k j}}{h_{k} h_{j}} \frac{\partial h_{k}}{\partial x_{j}}-\frac{E_{j j j}}{h_{k} h_{j}} \frac{\partial h_{i}}{\partial x_{k}}+\frac{E_{j i k}}{h_{i} h_{j}} \frac{\partial h_{i}}{\partial x_{j}}-\frac{E_{j j k}}{h_{i} h_{j}} \frac{\partial h_{j}}{\partial x_{i}}\right. \\
& \left.+\frac{E_{i l k}}{h_{j} h_{l}} \frac{\partial h_{j}}{\partial x_{l}}+\frac{E_{i 2 k}}{h_{j} h_{2}} \frac{\partial h_{j}}{\partial x_{?}}+\frac{E_{i 3 k}}{h_{j} h_{3}} \frac{\partial h_{j}}{\partial x_{3}}-\frac{E_{i j k}}{h_{j} h_{j}} \frac{\partial h_{j}}{\partial x_{j}}\right] \tag{A-8}
\end{align*}
$$

## Axisymmetric Case

When particularizing to the axisymmetric coordinate system of interest here, the above general forms are simplified considerably. For the axisymmetric case we take a cylindrical coordinate system defined by

$$
\begin{array}{ll}
x_{1}=x & h_{1}=1 \\
x_{2}=r & h_{2}=1 \\
x_{3}=\theta & h_{3}=r \tag{A-9}
\end{array}
$$

We may then write the forms discussed previously as follows: Gradient of a Scalar:

$$
\begin{equation*}
\phi, t=\frac{\partial \phi}{\partial t} \quad \phi_{, 1}=\frac{\partial \mu}{\partial x} \quad \phi_{, 2}=\frac{\partial \phi}{\partial r} \quad \phi, 3=\frac{1}{r} \frac{\partial \phi}{\partial \theta} \tag{A-10}
\end{equation*}
$$

## Gradient of a vector:

$$
\begin{equation*}
A_{i, j}=\frac{1}{h_{j}} \frac{\partial A_{i}}{\partial x_{j}}+\delta_{j 3} \delta_{i 3} \frac{A_{2}}{r}-\delta_{i 2} \delta_{3 j} \frac{A_{3}}{r} \tag{A-11}
\end{equation*}
$$

## Divergence of a vector

$$
\begin{equation*}
A_{j, j}=\sum_{j} \frac{1}{h_{j}} \frac{\partial A_{j}}{\partial x_{j}}+\frac{A_{2}}{r} \tag{A-12}
\end{equation*}
$$

## Directional derivative of a vector

$$
\begin{equation*}
B_{j} A_{i, j}=\sum_{j} \frac{B_{j}}{h_{j}} \frac{\partial A_{i}}{\partial x_{j}}+\delta_{i 3} \frac{B_{3} A_{2}}{r}-\delta_{i 2} \frac{B_{3} A_{3}}{r} \tag{A-13}
\end{equation*}
$$

## "Tensor directional derivative"

$$
\begin{equation*}
B_{k j} A_{i, j}=\sum_{j} \frac{B_{k j}}{h_{j}} \frac{\partial A_{i}}{\partial x_{j}}+\delta_{i 3} \stackrel{h_{j}}{=}, \quad \delta_{i 2} \frac{B_{k 3} A_{3}}{r} \tag{A-14}
\end{equation*}
$$

## Divergence of 2 nd order tensor

$$
\begin{equation*}
B_{i j, j}=\sum_{j} \frac{1}{h_{j}} \frac{\partial B_{i j}}{\partial x_{j}}+\frac{1}{r}\left\{B_{i 2}+\delta_{i 3} B_{23}-\delta_{i 2} B_{33}\right\} \tag{A-15}
\end{equation*}
$$

Divergence of 3 rd order tensor

$$
\begin{equation*}
E_{i j k, j}=\sum_{j} \frac{1}{h_{j}} \frac{\partial E_{i j k}}{\partial x_{j}}+\frac{1}{r}\left[\delta_{k 3} E_{i 32}-\delta_{k 2} E_{i 33}+\delta_{i 3} E_{23 k}-\delta_{i 2} E_{33 k}+E_{i 2 k}\right] \tag{A-16}
\end{equation*}
$$

The above forms were used, in conjunction with the conditions; $⿷ \equiv 0$ and $\frac{\partial(\bar{C})}{\partial \theta}=0$ deriving the axisymmetric equations presented in the body of this note.

## APPENDIX B <br> The Relationship Between The Reynolds And <br> The Mass-Averaged Dependent Variables

In the past, the Reynolds system of dependent variables has been used in both analytical investigations and interpretation of experimental data. Since the equations discussed in this report are cast in terms of the mass-averaged variables introduced by Favre, it is imfcrtant ir relating past work to this report to know how the variables, particularly the flisctuating variables, are related in the two systems.

We begin by summarizing the definition of the dependent variables in the two systems.

| Dependent Variable | Reynolds | Mass Averaged |
| :---: | :---: | :---: |
| u | $\bar{u}+u^{\prime}$ | $\tilde{u}^{+} u^{\prime \prime}$ |
| v | $\bar{v}+v^{\prime}$ | $v^{\prime}+v^{\prime \prime}$ |
| W | $\bar{w}+w^{\prime}$ | $\check{\mathrm{w}}+\mathrm{w}^{\prime \prime}$ |
| h | $\bar{l}+h^{\prime}$ | $\overline{\mathrm{h}}+\mathrm{h}^{\prime \prime}$ |
| T | $\bar{T}+\mathrm{T}^{\prime}$ | $\tilde{r}+\mathrm{T}^{\prime \prime}$ |
| 1 | $\bar{\mu}+\mu^{\prime}$ | $\tilde{\mu}+\mu^{\prime \prime}$ |
| $\rho$ | $\bar{p}+p^{\prime}$ | $\bar{\rho}+\rho^{\prime}$ |
| p | $\bar{p}+p^{\prime}$ | $\bar{p}+p^{\prime}$ |

The tilde indicates a mass-average defined in terms of the conventional time average as

$$
\begin{equation*}
\bar{a} \equiv \overline{p a} \tag{B-1}
\end{equation*}
$$

We also note that the variables $\rho$ and $p$ are defined identically in terms of mean and fluctuating components in bnth systems.

To reveal the relationship between the systems we examine the characteristics of an arbitrary variable $\alpha$, i.e.

$$
\begin{equation*}
\alpha=\bar{\alpha}+\alpha^{\prime}=\tilde{\alpha}+\alpha^{\prime \prime} \tag{B-2}
\end{equation*}
$$

srst, we note that in terms of the mass-averaged quantities

$$
\begin{equation*}
\overline{\rho \alpha}=\bar{\rho} \bar{\alpha}+\overline{\rho a^{\prime \prime}}=\overline{\rho \alpha}+\overline{\rho a^{\prime \prime}} \tag{B-3}
\end{equation*}
$$

and, thus,

$$
\begin{equation*}
\overline{\rho a^{\prime \prime}}=0 \tag{B-4}
\end{equation*}
$$

Further, we may evaluate $\overline{a^{\prime \prime}}$ from the above, since

$$
\begin{equation*}
0=\overline{\rho a^{\prime \prime}}=\overline{\rho a^{\prime \prime}}+\overline{\rho^{\prime} a^{\prime \prime}} \tag{B-5}
\end{equation*}
$$

and, thus,

$$
\begin{equation*}
\overline{a^{\prime \prime}}=-\frac{\overline{\rho^{\top} a^{\prime \prime}}}{\bar{j}} \tag{B-6}
\end{equation*}
$$

Note that, although $\overline{a^{\prime}}=0$ in the Reynolds system, $\overline{\alpha^{\prime \prime}} \neq 0$ for the massaveraged system.

In addition, since

$$
\begin{equation*}
\alpha=\bar{\alpha}+\alpha^{\prime}=\bar{\alpha}+\alpha^{\prime \prime} \tag{B-7}
\end{equation*}
$$

we can time average this to show

$$
\begin{equation*}
\bar{\alpha}-\tilde{\alpha}=\overline{\alpha^{\prime \prime}} \tag{B-8}
\end{equation*}
$$

and then show that

$$
\begin{align*}
\alpha^{\prime \prime} & =\bar{\alpha}+\alpha^{\prime}-\tilde{\alpha} \\
& =\alpha^{\prime}+\overline{\alpha^{\prime \prime}} \\
& =\alpha^{\prime}-\frac{\overline{\rho^{\prime}} \alpha^{\prime}}{\bar{\alpha}} \tag{B-9}
\end{align*}
$$

Muitiplyins this last equation by $\rho$ and time averaging, i.e.

$$
0=\overline{\rho a^{\prime \prime}}=\overline{\rho \alpha^{\prime}}-\frac{\overline{\rho^{\prime} \alpha^{\prime \prime} \rho}}{\bar{\rho}}=\overline{\rho^{\prime} \alpha^{\prime}}-\overline{\rho^{\prime} \alpha^{\prime \prime}}
$$

we see thet

$$
\begin{equation*}
\overline{\rho^{\prime} a^{\prime \prime}}=-T \tag{B-10}
\end{equation*}
$$

and, therefore,

$$
\begin{equation*}
\alpha^{\prime \prime}=\alpha^{\prime}-\frac{\overline{\rho^{\prime} \alpha^{\prime}}}{\bar{p}} \tag{B-11}
\end{equation*}
$$

Nith equation ( $B-1 i$ ), we may immediately deduce the relationship between turbulent transport of momentum and enthalpy expressed in the two systems of dependent variables. For momentum transport in the mass-averaged system, we require terms of the form

$$
\rho \alpha^{\prime \prime} B^{\prime \prime}
$$

which can be expressed in the reynolds system as follows:

$$
\begin{align*}
\overline{\rho \alpha^{\prime \prime} \beta^{\prime \prime}} & =\overline{\rho\left(\alpha^{\prime}-\frac{\rho^{\prime} \alpha^{\prime}}{\bar{\rho}}\right)\left(\beta^{\prime}-\frac{\rho^{\top} \beta^{\top}}{\delta}\right)} \\
& =\overline{\rho \alpha^{\prime} \beta^{\prime}}-\frac{\overline{\rho^{\prime} \alpha^{\prime} \rho^{\prime} \beta^{\prime}}}{\bar{\rho}} \tag{B-12}
\end{align*}
$$

We may now examine some specific examples. Consider the turbulent shear stress term $\overline{\rho u^{\prime \prime} v^{\prime \prime}}$ :

$$
\begin{equation*}
\overline{\rho u^{\prime \prime} v^{\prime \prime}}=\overline{\rho u^{\prime} v^{\prime}}-\frac{\overline{\rho^{\prime} u^{\prime} \rho^{\prime} v^{\prime}}}{\rho} \tag{B-13}
\end{equation*}
$$

similarly, the normal stress term $\overline{p u{ }^{\prime 2}}$ is just

$$
\begin{equation*}
\overline{\rho u^{\prime \prime 2}}=\overline{\rho u^{\prime 2}}-\frac{\left(\overline{\rho^{\prime} u^{\prime}}\right)^{2}}{\bar{\rho}} \tag{B-14}
\end{equation*}
$$

The turbulent heat transfer terms are as follows

$$
\begin{equation*}
\overline{\rho u^{\prime} h^{\prime \prime}}=\overline{\rho u^{\prime} h^{\prime}}-\frac{\overline{\rho^{\prime} u^{\prime}} \overline{\rho^{\prime} h^{\prime}}}{\bar{\rho}} \tag{B-15}
\end{equation*}
$$

and

$$
\begin{equation*}
\overline{\rho v^{\prime \prime} h^{\prime \prime}}=\overline{\rho v^{\prime} h^{\prime}}-\frac{\overline{\rho^{\prime} v^{\prime}} \overline{\rho^{\prime} h^{\prime}}}{\bar{\rho}} \tag{B-16}
\end{equation*}
$$

## REFERENCES

1. MacCormack, R. W.: Numerica! Solution of the Interacti.n if A Shock Wave with a Laminar Boundary Layer. Lecture Notes in Physics. "p. sceedings of the Second International Conference on Numerical Methods ir. Fluid Dynamics", Sept. 15-19, 1970. Edited by M. Holt, Springer Verlag.
2. Carter, J. E.: Numerical Solutions of The Navier-Stckes Equations Fcr The Supersoric Laminar Flow Over A Two Dimensional Compression Corner. NASA TR R-385, July 1972.
3. Favre, A.: Equations des Gaz Turbulents Compressibles. J. Mecan., Vol. 4, No. 3, Sept. 1965, pp. 361-390.
4. Donaldson, C. duP., and Sullivan, R. D.: an Invariant Second-Order Closure Model of the Compressible Turbulent Boundary Layer o:a a Flat Plate. A.R.A.P. Report No. 178, June 1972.
5. Laufer, J.: Thoughts on Compressible Turbulent Boundary Lavers, Symposium on Compressible Turbulent Boundary Layers, Langiey Research Cent 3 , Dec. 10-11, 1968, NASA SP-216.
6. Wilcox, D. C., and Alber, I. E.: A Turbulence Mrdel for High Speed Flows. Proceedings of the 1977 Heat Transfer and Fluid Mechanics Institute, San Fernando Valley State College, June l. -16, 1972.
7. Rose, W. C., "The Behavior of a Compressible Turbulent Bourdary Layer in a Shock-Wave Induced Adverse Pressure Gradient,' Ph.D. Thesis, Univ. of Washington, 1972. (Available as NASA TN D-7(192.)
8. Rotta, J.: Statiatical Theory of Inhomogene:ous Turbulence, Part. 1. Translation of "Statistische Theorie nichthomogener Turbulenz, 1. Mitteilung," Zeitschrift fuer Physik, Vol. 129, 1951, up. 547-572. NASA TT F-14, 560.
9. Townsend, A. A.: The Structure of Turbulent Shear Flow, Cambridge Liniversity Press, 1956.
10. Hayes, W. D.: The three-dimensional Boundary Layer. NAVORD Report 1313, NGTS 384, May 9, 1951.
