

The continuous strength method for the design of circular hollow sections

Craig Buchanan^{a,*}, Leroy Gardner^a, Andrew Liew^b

^a*Department of Civil and Environmental Engineering, Imperial College London, London, UK*

^b*Department of Civil, Environmental and Geomatic Engineering, Swiss Federal Institute of Technology (ETH), Zurich, Switzerland*

Abstract

Circular hollow sections (CHS) are widely used in a range of structural engineering applications. Their design is covered by all major design codes, which currently use elastic, perfectly-plastic material models and cross-section classification to determine cross-section compressive and flexural resistances. Experimental data for stocky sections show that this can result in overly conservative estimates of cross-section capacity. The continuous strength method (CSM) has been developed to reflect better the observed behaviour of structural sections of different metallic materials. The method is deformation based and allows for the rational exploitation of strain hardening. In this paper, the CSM is extended to cover the design of non-slender and slender structural steel, stainless steel and aluminium CHS, underpinned by and validated against 342 stub column and bending test results. Comparisons with the test results show that, overall, the CSM on average offers more accurate and less scattered predictions of axial and flexural capacities than existing design methods.

Keywords: circular hollow sections, continuous strength method, cross-section classification, local buckling, strain hardening

*Corresponding author

Email address: craig.buchanan08@imperial.ac.uk (Craig Buchanan)

1. Introduction

Circular hollow sections (CHS) have been manufactured and used in structures since the early 1800s as columns, beams, tension members and truss elements [1]. They have become increasingly attractive to designers due to their aesthetic appearance and their benefits over open sections such as superior torsional resistance, bi-axial bending resistance, reduced drag and loading in a fluid, ability to be filled with concrete to form a composite section and their reduced maintenance requirements with a smaller external area exposed to corrosive environments [1]. CHS are primarily thin-walled structural elements, and therefore local buckling, whether prior or subsequent to material yielding, is a primary consideration in their design.

1.1. Traditional CHS design methods

Current design codes use the concept of cross-section classification to separate circular hollow sections into discrete classes depending upon their susceptibility to local buckling. Four classes of cross-section are considered in EN 1993-1-1 [2] and BS 5950-1 [3] for structural steelwork, EN 1993-1-4 [4] for stainless steel and EN 1999-1-1 [5] for aluminium. In bending, class 1 cross-sections can reach and maintain their full plastic moment capacity M_{pl} with suitable rotation capacity for plastic design. Class 2 cross-sections are also capable of reaching their full plastic moment capacity but with limited rotation capacity. There is no equivalent to class 2 cross-sections in the AISC 360 [6] and AS 4100 [7] structural steel codes. Class 3 cross-sections are unable to reach their plastic moment capacity due to local buckling and their bending capacity is limited to the elastic moment capacity M_{el} . Class 4 cross-sections experience local buckling before reaching their elastic moment capacity, and are typically referred to as slender. In terms of axial resistance, the class 3 limit separates the non-slender cross-sections that are fully effective in compression (i.e. classes 1-3) from those that fail by local buckling before reaching their yield load (i.e. class 4). These traditional design methods also limit the maximum stress in the cross-section to the yield strength

30 f_y , neglecting the beneficial effects of strain hardening in metallic materials. Ex-
perimental results have shown that cross-section classification and limiting the
maximum stress to the yield stress can be overly conservative in estimating the
resistance of stocky (class 1-3, non-slender) cross-sections [8]. It is therefore
apparent that there are structural efficiency improvements to be sought over
35 existing design methods for CHS.

1.2. *The continuous strength method*

The continuous strength method (CSM) has been developed in recent years
to reflect better the observed characteristics of metallic structural elements.
40 Cross-section classification is replaced with a continuous relationship between
cross-section slenderness and deformation capacity (referred to in Section 2.5 as
the base curve), reflecting the continuous nature of cross-section capacity vary-
ing with local slenderness. A strain hardening material model is also adopted,
representing the behaviour seen in material tests, with an increase in strength
45 above the yield strength under plastic deformation.

The CSM has previously been developed for structural steel [8–10], stainless
steel [11] and aluminium [12] plated cross-sections, such as I-sections, square
hollow sections (SHS) and rectangular hollow sections (RHS) in compression
50 and bending, and also under combined bending [13]. The previous work has
shown that the CSM predicts enhanced capacities over existing methods; for
example, in the case of stainless steel, average enhancements in compressive
and bending resistances of 12% and 19% respectively were found [11]. The
natural progression is to extend the application of the CSM to circular hollow
55 sections, which is the focus of the present paper that builds upon prior work [14],
and the development process is described herein.

2. Extension of the CSM to CHS design

The process of extending the CSM to cover the design of CHS requires: i) the identification of the yield slenderness limit (i.e. the local slenderness limit below which significant benefit from strain hardening can be derived for non-slender cross-sections); ii) the formulation of the CSM non-slender and slender base curves defining the relationship between cross-section slenderness and deformation capacity; iii) the selection of appropriate material models; and iv) the derivation of resistance functions.

2.1. Cross-section slenderness

The local cross-section slenderness $\bar{\lambda}_c$ is defined in non-dimensional form by Eq. 1,

$$\bar{\lambda}_c = \sqrt{\frac{f_y}{\sigma_{cr}}} \quad (1)$$

where f_y is the material yield strength and σ_{cr} is the elastic critical buckling stress, which for a CHS in compression is calculated using Eq. 2,

$$\sigma_{cr} = \frac{E}{\sqrt{3(1-\nu^2)}} \frac{2t}{D} \quad (2)$$

where E is the Young's modulus, ν is the Poisson's ratio, D is the outer diameter and t is the wall thickness of the CHS.

Timoshenko [15] suggested that the local buckling stress in bending can be taken as 1.4 times that in compression based on experimental results [16], which effectively makes a cross-section in bending more stocky than the same cross-section in compression. Gerard and Becker [17] proposed a factor of 1.3 based upon the findings of Flügge [18]. However more recent analytical work [19–21] has showed that the maximum critical stress in bending is equal to the critical compressive stress for practical cylinder lengths. Differences also exist between international design codes in their treatment of compression and bending for CHS. Gardner et al. [22] noted that EN 1993-1-1 [2] and EN 1999-1-1 [5] utilise the same class 3 limits for both compression and bending, in contrast to BS

5950-1 [3], EN 1993-1-4 [4], AISC 360 [6] and AS 4100 [7] where different limits
80 are used. Utilising different slenderness limits in compression and bending is
equivalent to applying a factor to the local buckling stress. Given the findings
of the more recent research [19–21] and the conservative nature of the choice,
the elastic critical buckling stress in bending will be taken to be the same as
that in compression (see Eq. 2) in the present study and within the CSM.

85 *2.2. CHS experimental database*

A dataset of 342 experimental results on CHS in compression or bending has
been collated from the literature. The dataset includes stub column test re-
sults for hot-finished structural steel [23–27], very high strength structural steel
[28, 29], cold-formed structural steel [25, 26, 28, 30–47], austenitic stainless steel
90 [48–57], duplex stainless steel [51, 58, 59], ferritic stainless steel [60] and alu-
minium [61, 62], and four-point bending test results for hot-finished structural
steel [63–66], very high strength structural steel [67], cold-formed structural steel
[34, 63, 66, 68–73], austenitic stainless steel [48, 74, 75], duplex stainless steel
[74] and aluminium [76, 77]. Note that the very high strength structural steel
95 had a typical yield stress f_y around 1300 MPa [29]. The number of experimental
results used in the definition and assessment of the various aspects of the CSM
for CHS, i) the yield slenderness limit (see Section 2.3); ii) the base curve for
non-slender sections (see Section 2.5); iii) the base curve for slender sections (see
Section 2.5); and iv) the assessment of the capacity predictions (see Section 4)
100 are shown in Tables 1 and 2 for compression and bending respectively. The
number of specimens used in the different stages of the extension of the CSM
to CHS sometimes varies since not all required parameters were reported in the
literature.

105 *2.3. Yield slenderness limit*

The limiting local slenderness that delineates the transition between slender
and non-slender cross-sections needs to be defined. Above this limit there is no

Material	Yield derness limit	Base curve		CSM and code ca- pacity predictions	
		Non- slender sections	Slender sections	Non- slender sections	Slender sections
Hot-finished struc- tural steel	11	8	-	9	-
Very high strength structural steel	20	1	14	1	19
Cold-formed struc- tural steel	131	48	50	44	52
Stainless steel (to- tal)	76	31	26	39	35
Austenitic stainless steel	48	16	13	24	22
Duplex stainless steel	21	10	11	10	11
Ferritic stainless steel	7	5	2	5	2
Aluminium	15	7	7	7	8

Table 1: Number of CHS stub column test results used in the development of the various aspects of the CSM.

Material	Yield derness limit	Base curve		CSM and code ca- pacity predictions	
		Non- slender sections	Slender sections	Non- slender sections	Slender sections
Hot-finished struc- tural steel	14	2	-	14	-
Very high strength structural steel	12	1	8	1	11
Cold-formed struc- tural steel	44	9	17	13	27
Stainless steel (to- tal)	12	1	8	3	9
Austenitic stainless steel	8	1	4	3	5
Duplex stainless steel	4	-	4	-	4
Ferritic stainless steel	-	-	-	-	-
Aluminium	7	3	4	3	4

Table 2: Number of CHS four-point bending test results used in the development of the various aspects of the CSM.

significant benefit from strain hardening with the cross-section buckling locally below the yield load in compression or elastic moment in bending. This limit is identified by plotting the ultimate capacity of the stub columns normalised by their yield load (N_u/N_y) against cross-section slenderness $\bar{\lambda}_c$, defined by Eq. 1, as shown in Fig. 1. A linear regression fit can then identify the limiting local slenderness where the ultimate axial load equals the yield load, which from Fig. 1 is $\bar{\lambda}_c \approx 0.40$. The class 3 limits from current design codes are also plotted in Fig. 1, and it can be seen that the identified limiting local slenderness is compatible with the class 3 limit for aluminium given in EN 1999-1-1 [5]; however it is above the existing structural steel and stainless steel class 3 limits. There is also some scatter in the stub column dataset. Consequently, a lower value of $\bar{\lambda}_c = 0.3$ for the yield slenderness limit is proposed as this represents approximately a lower bound to the assembled dataset and is generally comparable with existing codes.

It is evident from Fig. 1 that there are no clear discontinuities in the dataset and that limiting the maximum material stress to the yield stress is overly conservative for stocky cross-sections. The ultimate bending moments from the collected beam tests normalised by the elastic moment (M_u/M_{el}) are plotted in Fig. 2 against the cross-section slenderness. As for compression, in contrast to the EN 1993-1-1 [2] resistance line, there are no apparent discontinuities in the bending capacity dataset that cross-section classification would otherwise suggest and again limiting the material stress to the yield stress can lead to under-predictions of the ultimate cross-section capacity for stocky sections. The results show that the previous yield slenderness limit of $\bar{\lambda}_c = 0.3$ can also be applied for bending.

2.4. Normalised deformation capacity (strain ratio)

In the CSM, cross-section classification is replaced by a continuous relationship between local slenderness and deformation capacity. This deformation capacity is called the strain ratio ($\varepsilon_{csm}/\varepsilon_y$) and is defined as the strain at ultimate load

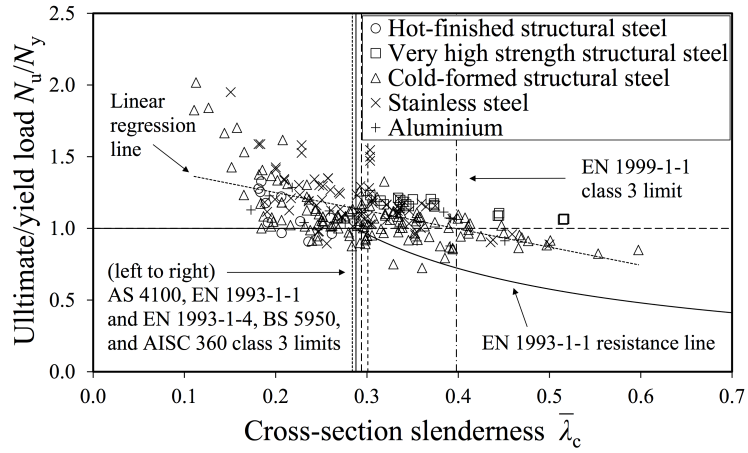


Figure 1: Normalised ultimate axial resistance N_u/N_y varying with local slenderness $\bar{\lambda}_c$.

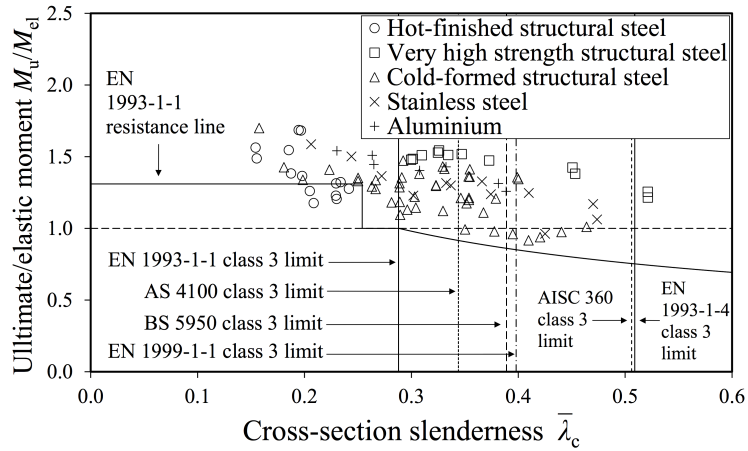


Figure 2: Normalised ultimate flexural resistance M_u/M_{el} varying with local slenderness $\bar{\lambda}_c$.

ε_{csm} , which is taken as the limiting strain for the cross-section, normalised by the yield strain ε_y . The strain ratio is determined from stub column and four-point bending experiments, as described below.

2.4.1. Axial compression

For stub columns with a local slenderness below the yield slenderness limit ($\bar{\lambda}_c \leq 0.3$) and where the ultimate load exceeds the yield load ($N_u \geq N_y$), the strain ratio is expressed as a function of the strain at ultimate load divided by the yield strain ($\varepsilon_{lb}/\varepsilon_y$) (Eqs. 3 and 4). The strain at ultimate load ε_{lb} can be calculated from the initial specimen length L and the end-shortening δ_u at the ultimate load N_u . A 0.002 strain offset is subtracted from ε_{lb} for materials with a rounded stress-strain response for compatibility with the material models adopted in Section 2.6, as shown in Eq. 3.

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For $\bar{\lambda}_c \leq 0.3$, $N_u \geq N_y$ and a rounded material response:

$$\frac{\varepsilon_{csm}}{\varepsilon_y} = \frac{\varepsilon_{lb} - 0.002}{\varepsilon_y} = \frac{\delta_u/L - 0.002}{\varepsilon_y} \quad (3)$$

For $\bar{\lambda}_c \leq 0.3$, $N_u \geq N_y$ and a sharply defined yield point:

$$\frac{\varepsilon_{csm}}{\varepsilon_y} = \frac{\varepsilon_{lb}}{\varepsilon_y} = \frac{\delta_u/L}{\varepsilon_y} \quad (4)$$

For specimens that do not exceed the yield load ($N_u < N_y$) or that have slender cross-sections ($\bar{\lambda}_c > 0.3$), the strain ratio is determined as the ratio of the ultimate load to yield load (N_u/N_y), as given by Eq. 5.

For $\bar{\lambda}_c > 0.3$ or $N_u < N_y$

$$\frac{\varepsilon_{csm}}{\varepsilon_y} = \frac{N_u}{N_y} \quad (5)$$

2.4.2. Four-point bending

Under uniform bending, the strain can be determined as the product of the curvature κ and the distance from the elastic neutral axis y (Eq. 6); the curvature at the ultimate moment and the elastic moment are termed κ_u and κ_{el}

respectively.

$$\varepsilon = \kappa y \quad (6)$$

κ_u can be calculated from existing four-point bending experimental data using Eq. 7 [78].

$$\kappa = \frac{1}{r} = \frac{8(D_M - D_L)}{4(D_M - D_L)^2 + L_2^2} \quad (7)$$

where r is the radius of curvature, D_M and D_L are the displacements at the midspan and loading points respectively and L_2 is the length of the central region of the beam between the two loading points. The curvature at the elastic moment κ_{el} can be determined from Eq. 8, where I is the second moment of area.

$$\kappa_{el} = \frac{M_{el}}{EI} \quad (8)$$

The strain ratio for four-point bending is similar in principle to axial compression, being defined as a function of the maximum strain in the cross-section at the ultimate moment normalised by the yield strain ($\varepsilon_{lb}/\varepsilon_y$) for $\bar{\lambda}_c \leq 0.3$ and where the ultimate moment exceeds the elastic moment ($M_u \geq M_{el}$) (Eqs. 9 and 10). The 0.002 offset is again subtracted for materials with a rounded
160 stress-strain response as shown in Eq. 9, where y_{max} is the distance from the elastic neutral axis to the extreme fibre of the cross-section.

For $\bar{\lambda}_c \leq 0.3$, $M_u \geq M_{el}$ and a rounded material response:

$$\frac{\varepsilon_{csm}}{\varepsilon_y} = \frac{\varepsilon_{lb} - 0.002}{\varepsilon_y} = \frac{\kappa_u y_{max} - 0.002}{\kappa_{el} y_{max}} \quad (9)$$

For $\bar{\lambda}_c \leq 0.3$, $M_u \geq M_{el}$ and a sharply defined yield point:

$$\frac{\varepsilon_{csm}}{\varepsilon_y} = \frac{\varepsilon_{lb}}{\varepsilon_y} = \frac{\kappa_u y_{max}}{\kappa_{el} y_{max}} \quad (10)$$

If the cross-section is slender ($\bar{\lambda}_c > 0.3$) or the ultimate moment is less than the
165 elastic moment ($M_u < M_{el}$), the strain ratio is taken as the ultimate moment normalised by the elastic moment (M_u/M_{el}), as shown in Eq. 11.

For $\bar{\lambda}_c > 0.3$ or $M_u < M_{el}$

$$\frac{\varepsilon_{csm}}{\varepsilon_y} = \frac{M_u}{M_{el}} \quad (11)$$

2.5. Base curve

The base curve defines the relationship between the deformation capacity of a cross-section (i.e. the maximum limiting strain that it can endure ε_{csm}) and the local slenderness of the cross-section. Base curves for non-slender ($\bar{\lambda}_c \leq 0.3$) and slender ($\bar{\lambda}_c > 0.3$) CHS with the forms of Eqs. 12 and 13 respectively can be fitted to the experimental strain ratios derived from the axial and bending results from the literature (Table 1), as plotted in Figs. 3 and 4.

$$\frac{\varepsilon_{csm}}{\varepsilon_y} = \frac{A}{\bar{\lambda}_c^B} \text{ for } \bar{\lambda}_c \leq 0.3 \quad (12)$$

$$\frac{\varepsilon_{csm}}{\varepsilon_y} = \left(1 - \frac{F}{\bar{\lambda}_c^G}\right) \frac{1}{\bar{\lambda}_c} \text{ for } \bar{\lambda}_c > 0.3 \quad (13)$$

Eq. 12 is consistent with previous implementations of the CSM (refer to Section 1.2), and is similar in form to the relationship between normalised elastic buckling strain $\varepsilon_{cr}/\varepsilon_y$ and local slenderness $\bar{\lambda}_c$, given by Eq. 14. Eq. 13 is of the same general form as the normalised strength curves from the Direct Strength Method (DSM) [79].

$$\frac{\varepsilon_{cr}}{\varepsilon_y} = \frac{1}{\bar{\lambda}_c^2} \quad (14)$$

The two chosen base curves, given by Eq. 15 for non-slender CHS and by Eq. 16 for slender CHS, generally represent a lower bound to the dataset and meet at the yield slenderness limit ($\bar{\lambda}_c = 0.3$) previously identified at a strain ratio $\varepsilon_{csm}/\varepsilon_y$ of unity.

$$\frac{\varepsilon_{csm}}{\varepsilon_y} = \frac{4.44 \times 10^{-3}}{\bar{\lambda}_c^{4.5}} \text{ for } \bar{\lambda}_c \leq 0.3 \text{ but } \frac{\varepsilon_{csm}}{\varepsilon_y} \leq \min\left(15, \frac{C_1 \varepsilon_u}{\varepsilon_y}\right) \quad (15)$$

$$\frac{\varepsilon_{csm}}{\varepsilon_y} = \left(1 - \frac{0.224}{\bar{\lambda}_c^{0.342}}\right) \frac{1}{\bar{\lambda}_c^{0.342}} \text{ for } 0.3 < \bar{\lambda}_c \leq 0.6 \quad (16)$$

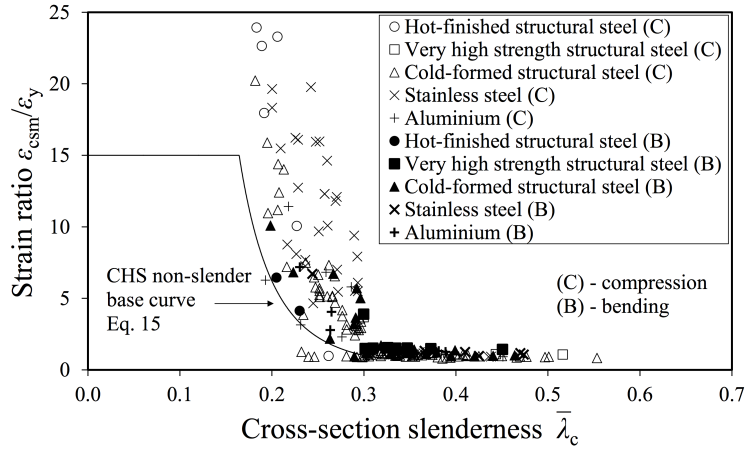


Figure 3: CHS CSM non-slender base curve with collected experimental data.

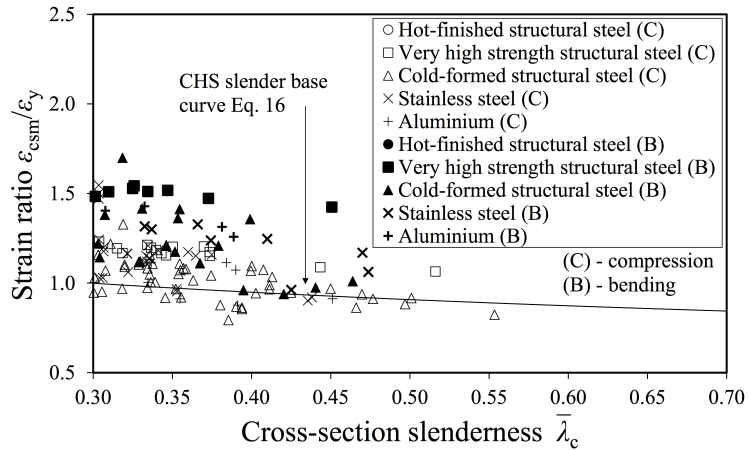


Figure 4: CHS CSM slender base curve with collected experimental data.

170 Upper limits are placed upon the strain ratio, in effect limiting the extent to
 which the non-slender cross-sections can deform. The upper limit of 15 in Eq. 15
 is the material ductility requirement from EN 1993-1-1 [2] and is applied to all
 metallic materials. A second upper limit is applied to cold-formed and very
 high strength structural steel ($C_1 = 0.4$), austenitic and duplex stainless steel
 175 ($C_1 = 0.1$), ferritic stainless steel ($C_1 = 0.4$) and aluminium ($C_1 = 0.5$) to
 prevent over-predictions of cross-section resistance due to the chosen simplified
 material model. ε_u is the strain at the ultimate tensile stress of the material,
 and is discussed further in Section 2.6. An upper limit of $\bar{\lambda}_c \leq 0.6$ is placed
 upon the slender base curve since experimental data has not been examined
 180 beyond this slenderness.

2.6. Material models

An elastic, linear strain hardening material model (of slope E_{sh}), as shown in
 Fig. 5, is adopted in the CSM, replacing the traditional elastic, perfectly-plastic
 material model typically employed in existing design codes. The coefficients C_1
 to C_4 are defined in Section 2.6.5. The CSM limiting stress f_{csm} is defined
 by Eq. 17 for $\varepsilon_{csm}/\varepsilon_y < 1$ and by Eq. 18 for $\varepsilon_{csm}/\varepsilon_y \geq 1$. Eq. 18 represents
 the strain hardening behaviour of the material through the strain hardening
 modulus E_{sh} .

$$f_{csm} = E\varepsilon_{csm} \text{ for } \frac{\varepsilon_{csm}}{\varepsilon_y} < 1 \quad (17)$$

$$f_{csm} = f_y + E_{sh}\varepsilon_y \left(\frac{\varepsilon_{csm}}{\varepsilon_y} - 1 \right) \text{ for } \frac{\varepsilon_{csm}}{\varepsilon_y} \geq 1 \quad (18)$$

2.6.1. Hot-finished structural steel sections

For hot-finished structural steel sections, the strain hardening modulus E_{sh}
 proposed by Foster [10] has been utilised. This is the simplest model adopted as
 strain hardening is taken as zero (Eq. 19), reducing the stress-strain response to
 the traditional linear elastic, perfectly-plastic model. This is due to the extensive
 yield plateau associated with hot-finished tubes. Work is currently underway to
 develop a more refined material model for hot-finished structural steel sections

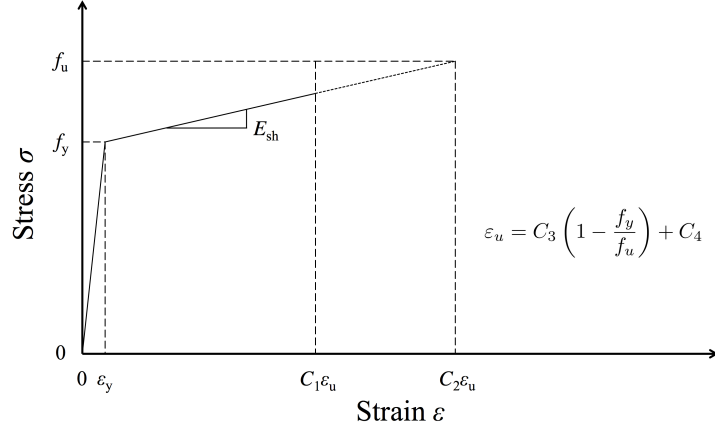


Figure 5: Elastic, linear strain hardening CSM material model.

that takes into account strain hardening at the end of the yield plateau, leading to strain hardening benefits for cross-sections with high deformation capacities.

$$\frac{E_{sh}}{E} = 0 \quad (19)$$

2.6.2. Cold-formed structural steel sections

For cold-formed structural steel sections, based on recent work by Gardner et al. [80], the strain hardening modulus E_{sh} is defined by Eqs. 20 and 21, depending on the ratio $\varepsilon_y/\varepsilon_u$. This model is also applied to the very high strength structural steel CHS considered herein.

For $\varepsilon_y/\varepsilon_u < 0.45$:

$$E_{sh} = \frac{f_u - f_y}{0.45\varepsilon_u - \varepsilon_y} \quad (20)$$

For $\varepsilon_y/\varepsilon_u \geq 0.45$:

$$E_{sh} = 0 \quad (21)$$

The material strain ε_u corresponding to the ultimate tensile stress f_u may be predicted using Eq. 22 [80].

$$\varepsilon_u = 0.6 \left(1 - \frac{f_y}{f_u} \right) \quad (22)$$

2.6.3. Stainless steel sections

For stainless steel sections, the material model developed by Afshan and Gardner [11] is utilised for the austenitic and duplex grades. The strain hardening modulus E_{sh} is predicted from Eq. 23 and ε_u is given by Eq. 24, which is taken from EN 1993-1-4 [4, 81]. The ultimate tensile stress f_u , if not provided, can be estimated using Eq. 25 [81].

$$E_{sh} = \frac{f_u - f_y}{0.16\varepsilon_u - \varepsilon_y} \quad (23)$$

$$\varepsilon_u = 1 - \frac{f_y}{f_u} \quad (24)$$

$$f_u = \frac{f_y}{0.2 + 185f_y/E} \quad (25)$$

The material model developed by Bock et al. [82] has been applied to the ferritic grades. This model is the same as that proposed for cold-formed structural steel sections, and therefore utilises Eqs. 20, 21 and 22 [82]. If f_u is not provided it can be predicted using Eq. 26 [83, 84].

$$f_u = \frac{f_y}{0.46 + 145f_y/E} \quad (26)$$

190 2.6.4. Aluminium sections

For aluminium alloy sections, the material model proposed by Su et al. [12] is adopted in this study. The predictive expression for the strain hardening modulus E_{sh} is given by Eq. 27. This is similar in form to the previous cold-formed structural steel and stainless steel models. The material ultimate strain ε_u may be predicted from Eq. 28. The latter expression is only applicable when the ratio of the ultimate stress f_u to yield stress f_y exceeds 1.01.

$$E_{sh} = \frac{f_u - f_y}{0.5\varepsilon_u - \varepsilon_y} \quad (27)$$

For $f_u/f_y > 1.01$

$$\varepsilon_u = 0.13 \left(1 - \frac{f_y}{f_u} \right) + 0.06 \quad (28)$$

	C_1	C_2	C_3	C_4
Hot-finished structural steel	- 1	- 1	- 1	- 1
Very high strength structural steel	0.40	0.45	0.60	0
Cold-formed structural steel	0.40	0.45	0.60	0
Austenitic and duplex stainless steel	0.10	0.16	1.00	0
Ferritic stainless steel	0.40	0.45	0.60	0
Aluminium	0.50	0.50	0.13	0.06

¹ Hot-finished structural steel material model under development

Table 3: Summary of coefficients for the CSM material model.

2.6.5. Summary of material models

The CSM provides a harmonised design approach across all metallic materials. A common base curve (Eqs. 15 and 16) is used for CHS of all materials, while the differing strain hardening characteristics of the various materials are accounted
195 for through different coefficients in the general elastic, linear hardening $\sigma - \varepsilon$ curve (see Fig. 5). There are four coefficients C_1 , C_2 , C_3 and C_4 . C_1 defines a ‘cut-off’ strain (in Eq. 15) to avoid over-predictions of material strength when using the elastic, linear hardening material model; C_2 is used in Eq. 29 to define the strain hardening slope E_{sh} ; and C_3 and C_4 are used in the predictive
200 expression for ultimate strain (Eq. 30), which is also needed to determine E_{sh} . Values of the coefficients for the different considered materials are summarised in Table 3.

$$E_{sh} = \frac{f_u - f_y}{C_2 \varepsilon_u - \varepsilon_y} \quad (29)$$

$$\varepsilon_u = C_3 \left(1 - \frac{f_y}{f_u} \right) + C_4 \quad (30)$$

3. CSM cross-section resistance functions for CHS

The cross-section resistances in compression or bending can now be determined
205 utilising the deformation capacity ($\varepsilon_{csm}/\varepsilon_y$) predicted from the base curve, together with the adopted material model.

3.1. Compressive resistance

The CSM axial compressive resistance for non-slender cross-sections N_{csm} is calculated as the product of the gross cross-section area A and the CSM limiting material stress f_{csm} , as given by Eq. 31.

$$N_{csm} = Af_{csm} \text{ for } \bar{\lambda}_c \leq 0.3 \quad (31)$$

The strength benefit for non-slender cross-sections arises when the CSM limiting stress f_{csm} exceeds the yield stress f_y . Consequently for hot-finished structural
210 steel sections there are no strength benefits due to the strain hardening model adopted in Eq. 19, while for the other considered materials, there is additional resistance with increasing deformation capacity.

For slender cross-sections, the axial compressive resistance can be determined using Eq. 32, which is the yield load Af_y factored by the strain ratio.

$$N_{csm} = \frac{\varepsilon_{csm}}{\varepsilon_y} Af_y \text{ for } 0.3 < \bar{\lambda}_c \leq 0.6 \quad (32)$$

3.2. Bending resistance

215 The derivation of the CSM bending resistance function for non-slender CHS [85] is first described. In the derivation, it is assumed that plane sections remain plane and normal to the neutral axis during bending, and that the cross-section shape does not significantly distort before the outer-fibre strain ε_{csm} is attained. The corresponding linear strain and bi-linear stress distributions (arising for the
220 elastic, linear hardening material model) for half of a CHS are shown in Fig. 6.

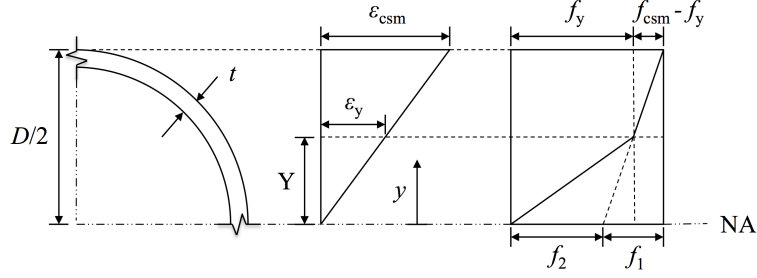


Figure 6: Strain and stress profiles for half of a CHS (a quarter of the cross-section is shown).

From Fig. 6, the bending capacity of a cross-section can be expressed by Eq. 33, in terms of the elastic section modulus W_{el} , plastic section modulus W_{pl} , and the introduced modulus W_w .

$$M_{csm} = W_{pl}f_{csm} - (W_{pl} - W_{el})f_1 - W_w f_2 \quad (33)$$

The term $W_w f_2$ represents the moment caused by the triangular shaped stress block associated with stress f_2 for $|y| \leq Y$. The first yield distance Y from the neutral axis NA is given by Eq. 34.

$$Y = \frac{0.5D}{\varepsilon_{csm}/\varepsilon_y} \quad (34)$$

Using the CSM material model (Eq. 18 from Section 2.6) and the stress distribution geometry in Fig. 6, stresses f_1 and f_2 are determined as follows.

$$\frac{f_1}{f_y} = \frac{E_{sh}}{E} \frac{\varepsilon_{csm}}{\varepsilon_y} \quad (35)$$

$$\frac{f_2}{f_y} = 1 - \frac{E_{sh}}{E} \quad (36)$$

Normalising Eq. 33 by the plastic moment capacity $M_{pl} = W_{pl}f_y$ and substituting in the expressions for f_{csm} , f_1 and f_2 gives Eq. 37.

$$\frac{M_{csm}}{M_{pl}} = 1 + \frac{E_{sh}}{E} \left(\frac{\varepsilon_{csm}}{\varepsilon_y} \frac{W_{el}}{W_{pl}} - 1 \right) - \frac{W_w}{W_{pl}} \left(1 - \frac{E_{sh}}{E} \right) \quad (37)$$

The moment $M_{f_2} = W_w f_2$ is defined in Eq. 38, from which the introduced modulus W_w can be determined using Eq. 39, where the function $g(y)$ represents the

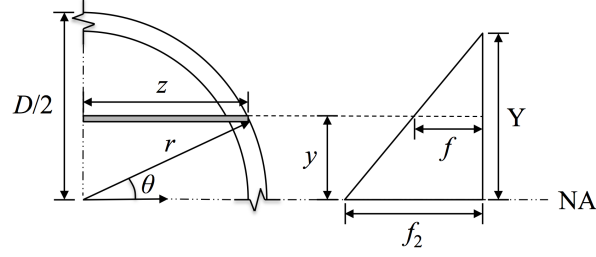


Figure 7: Geometry for the derivation of W_w for CHS.

triangular stress distribution normalised by f_2 . The integral is only evaluated to the first yield point $|y| = Y$, and the associated area to integrate over is A_Y .

$$M_{f_2} = W_w f_2 = \int_{A_Y} f y dA_Y = \int_{A_Y} f_2 g(y) y dA_Y \quad (38)$$

$$W_w = \int_{A_Y} g(y) y dA_Y \quad (39)$$

If integration is performed over one quarter of the cross-section, as in Fig. 7, then the result may be multiplied by four for the entire CHS to give the modulus W_w corresponding to the area A_Y . From the equation of a circle (Eq. 40), in z and y co-ordinates and with radius r , the positive branch length can be calculated from Eq. 41.

$$z^2 + y^2 = r^2 \quad (40)$$

$$z = r \sqrt{1 - \frac{y^2}{r^2}}. \quad (41)$$

The introduced modulus $W_{w,r}$ for a solid circle of radius r can be determined from Eq. 42, which is four times the integral of the stress triangle multiplied by the associated area and lever arm.

$$W_{w,r} = 4 \int_0^Y y \left(1 - \frac{y}{Y}\right) z dy = 4r \int_0^Y y \sqrt{1 - \frac{y^2}{r^2}} dy - \frac{4r}{Y} \int_0^Y y^2 \sqrt{1 - \frac{y^2}{r^2}} dy \quad (42)$$

where Y from Eq. 34 is the value of y at first yield, which is constant for a given

strain ratio. The solution to this integral is given by Eq. 43.

$$W_{w,r} = \frac{4r^3}{3} \left[1 - \left(1 - \frac{Y^2}{r^2} \right)^{\frac{3}{2}} \right] - \frac{r}{Y} \left[\frac{Y}{2} (2Y^2 - r^2) \sqrt{1 - \frac{Y^2}{r^2}} + \frac{r^3}{2} \sin^{-1} \left(\frac{Y}{r} \right) \right] \quad (43)$$

W_w for a circular hollow section can then be calculated from $W_w = W_{w,r_2} - W_{w,r_1}$, where W_{w,r_2} is evaluated for the outer radius $r_2 = D/2$, leading to Eq. 44, and W_{w,r_1} is determined for the inner radius $r_1 = r_2 - t$, leading to Eq. 45.

$$W_{r,r_2} = \frac{4r_2^3}{3} [1 - \cos^3 \alpha] - \frac{r_2^3}{\sin \alpha} \left[\sin^3 \alpha \cos \alpha + \frac{\alpha}{2} - \frac{\sin \alpha \cos \alpha}{2} \right] \quad (44)$$

$$W_{r,r_1} = \frac{4r_1^3}{3} [1 - \cos^3 \beta] - \frac{r_1^3}{\sin \beta} \left[\sin^3 \beta \cos \beta + \frac{\beta}{2} - \frac{\sin \beta \cos \beta}{2} \right] \quad (45)$$

where $\sin \alpha = \frac{Y}{r_2}$, $\cos \alpha = \sqrt{1 - \frac{Y^2}{r_2^2}}$, $\sin \beta = \frac{Y}{r_1}$ and $\cos \beta = \sqrt{1 - \frac{Y^2}{r_1^2}}$. The analytical formula is valid for $r_2/r_1 < \varepsilon_{csm}/\varepsilon_y$.

225 Substituting $W_w = W_{w,r_2} - W_{w,r_1}$ and the expressions for the two introduced moduli into Eq. 37 leads to the exact analytical CSM bending resistance function. However, due to the lengthy W_w term, the exact expression is not suitable for use in design, and therefore a simplified design equation is sought.

For a strain ratio $\varepsilon_{csm}/\varepsilon_y$ of unity, $f_{csm} = f_y$, and the introduced modulus W_w simplifies to:

$$W_w = W_{pl} - W_{el} \quad (46)$$

For strain ratios greater than unity, the introduced modulus W_w can be approximated by Eq. 47 [8], which can then be substituted into Eq. 37 to give the normalised CSM moment capacity expressed by Eq. 48.

$$W_w = (W_{pl} - W_{el}) / \left(\frac{\varepsilon_{csm}}{\varepsilon_y} \right)^2 \quad (47)$$

$$\frac{M_{csm}}{M_{pl}} = 1 + \frac{E_{sh}}{E} \left(\frac{\varepsilon_{csm}}{\varepsilon_y} \frac{W_{el}}{W_{pl}} - 1 \right) - \left(1 - \frac{W_{el}}{W_{pl}} \right) \left(1 - \frac{E_{sh}}{E} \right) / \left(\frac{\varepsilon_{csm}}{\varepsilon_y} \right)^2 \quad (48)$$

By noting that in general $E_{sh}/E \ll 1$, the $(1 - E_{sh}/E)$ term can conservatively be taken as unity as the final term is subtractive, while the remaining expression can be forced through M_{el} at a strain ratio of unity to give the simplified M_{csm} equation provided as Eq. 49.

$$M_{csm} = M_{pl} \left[1 + \frac{E_{sh}}{E} \frac{W_{el}}{W_{pl}} \left(\frac{\varepsilon_{csm}}{\varepsilon_y} - 1 \right) - \left(1 - \frac{W_{el}}{W_{pl}} \right) / \left(\frac{\varepsilon_{csm}}{\varepsilon_y} \right)^2 \right] \text{ for } \bar{\lambda}_c \leq 0.3 \quad (49)$$

230 Design equations of this format have also been used for determining the CSM bending resistance of I-sections and box sections [9]. The exact analytical CSM bending resistance expression, Eq. 37, is plotted along with the simplified CSM design expression, Eq. 49, in Fig. 8 for a typical ratio of E_{sh}/E of 1/100. The design equation can be seen to tend towards the analytical expression for the
 235 higher strain ratios where W_w has a smaller influence, and that, overall, the difference between the two curves is minimal.

The variation in bending capacity M_{csm}/M_{pl} with strain ratio $\varepsilon_{csm}/\varepsilon_y$ for various strain hardening ratios E_{sh}/E is plotted in Fig. 9. The bending capacity
 240 at a strain ratio of unity is the elastic moment M_{el} . The subsequent increase in bending capacity for $\varepsilon_{csm}/\varepsilon_y > 1$ is dependent upon the strain hardening ratio, where a larger strain hardening modulus leads to a greater increase in bending capacity with increasing strain ratio $\varepsilon_{csm}/\varepsilon_y$. If the strain hardening modulus is taken as zero, the bending capacity is asymptotic to the plastic moment ca-
 245 pacity. It should be noted that the CSM bending resistance function for CHS is sensitive to the strain ratio in the region of $\varepsilon_{csm}/\varepsilon_y < 3$, reflecting the relatively high shape factor of CHS and the marked increases in moment capacity that arise during the initial spread of plasticity.

250 For slender cross-sections the CSM bending resistance M_{csm} can be calculated by factoring the elastic moment capacity $W_{el}f_y$ by the strain ratio as shown in

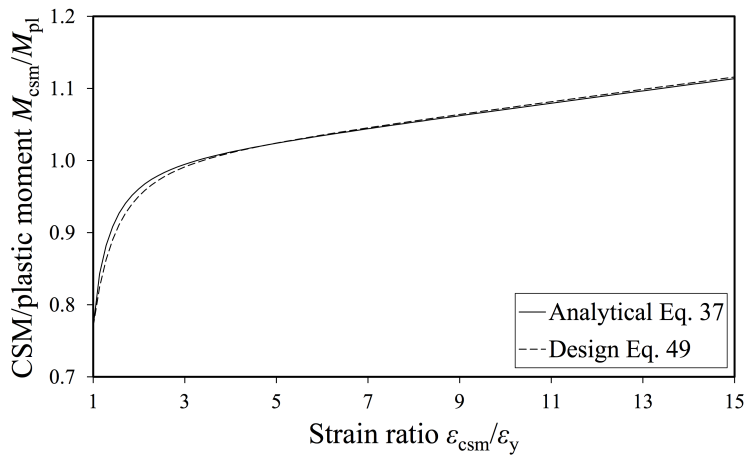


Figure 8: Comparison between exact and approximate (design) CSM bending capacity predictions M_{csm}/M_{pl} with varying $\varepsilon_{csm}/\varepsilon_y$.

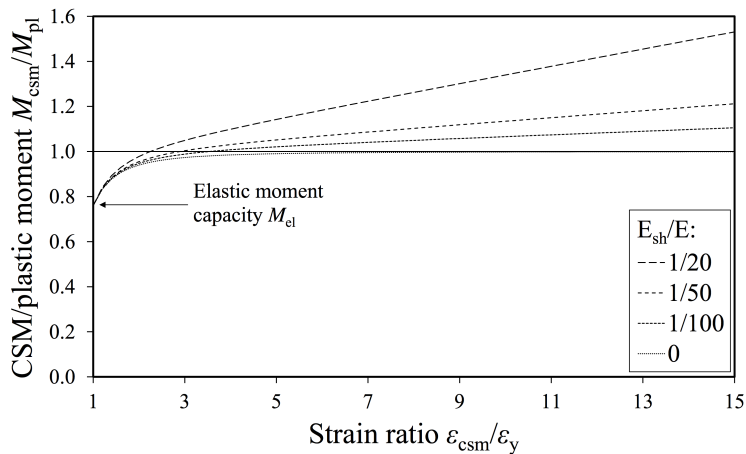


Figure 9: Influence of E_{sh}/E on M_{csm}/M_{pl} with varying $\varepsilon_{csm}/\varepsilon_y$.

Eq. 50.

$$M_{csm} = \frac{\varepsilon_{csm}}{\varepsilon_y} W_{el} f_y \text{ for } 0.3 < \bar{\lambda}_c \leq 0.6 \quad (50)$$

4. Comparison with test data and existing design methods

The CSM predictions for compression and bending resistances have been compared with the measured ultimate values from the collected experiments. The average ultimate test loads N_u and moments M_u normalised by the CSM (N_{csm} , M_{csm}) and Eurocode (N_{EC} , M_{EC}) predictions have been determined for each material type and are summarised in Tables 4 and 6 respectively for non-slender cross-sections ($\bar{\lambda}_c \leq 0.3$), and in Tables 5 and 7 for slender cross-sections ($\bar{\lambda}_c > 0.3$). EN 1993-1-1 [2] and EN 1993-1-4 [4] do not provide design expressions for slender CHS in compression or bending; the design formulae for slender cross-sections from BS 5950-1 [3] with updated Eurocode class 3 limits have therefore been utilised [86]. EN 1999-1-1 [5] provides design expressions for slender class 4 aluminium CHS resistances in compression and bending. The coefficients of variation (COV) have also been calculated to quantify and allow comparisons of the scatter of the predictions. The ultimate experimental loads normalised by their CSM and Eurocode predictions ($N_{u,pred}$, $M_{u,pred}$) have been plotted for compression in Fig. 10 and bending in Fig. 11.

Overall, on average, for both non-slender and slender cross-sections the CSM predicts cross-section compression and bending resistances that are more accurate and consistent compared with those from the Eurocodes. Looking initially at the non-slender predictions, apart from for hot-finished structural steel, the CSM predictions are on average closer to the measured ultimate resistance than those from the Eurocodes and exhibit reduced scatter. For hot-finished structural steel, the CSM and EN 1993-1-1 [2] predictions are the same for cross-sections in compression (due to the use of $E_{sh} = 0$), while in bending the CSM is more conservative than EN 1993-1-1 [2] as the predicted capacity does not

Material	Mean		COV	
	N_u/N_{csm}	N_u/N_{EC}	N_u/N_{csm}	N_u/N_{EC}
Hot-finished structural steel	1.08	1.08	0.12	0.12
Very high strength structural steel	1.19	1.21	- ¹	- ¹
Cold-formed structural steel	1.12	1.17	0.17	0.20
Stainless steel	1.19	1.26	0.12	0.16
Aluminium	1.09	1.13	0.08	0.08
Average	1.14	1.19	0.14	0.18

¹ Insufficient experimental data

Table 4: CSM and Eurocode compression resistance prediction comparison for $\bar{\lambda}_c \leq 0.3$.

Material	Mean		COV	
	N_u/N_{csm}	N_u/N_{EC}	N_u/N_{csm}	N_u/N_{EC}
Hot-finished structural steel	- ¹	- ¹	- ¹	- ¹
Very high strength structural steel	1.20	1.50	0.02	0.13
Cold-formed structural steel	1.04	1.27	0.10	0.14
Stainless steel	1.15	1.33	0.12	0.11
Aluminium	1.05	1.06	0.09	0.09
Average	1.10	1.31	0.12	0.15

¹ Insufficient experimental data

Table 5: CSM and Eurocode compression resistance prediction comparison for $0.3 < \bar{\lambda}_c \leq 0.6$.

Material	Mean		COV	
	M_u/M_{csm}	M_u/M_{EC}	M_u/M_{csm}	M_u/M_{EC}
Hot-finished structural steel	1.06	1.05	0.11	0.12
Very high strength structural steel	1.47	1.49	- ¹	- ¹
Cold-formed structural steel	1.10	1.18	0.10	0.12
Stainless steel	1.15	1.24	0.01	0.09
Aluminium	1.19	1.36	0.03	0.15
Average	1.11	1.16	0.11	0.15

¹ Insufficient experimental data

Table 6: CSM and Eurocode bending resistance prediction comparison for $\bar{\lambda}_c \leq 0.3$.

Material	Mean		COV	
	M_u/M_{csm}	M_u/M_{EC}	M_u/M_{csm}	M_u/M_{EC}
Hot-finished structural steel	- ¹	- ¹	- ¹	- ¹
Very high strength structural steel	1.50	1.63	0.05	0.04
Cold-formed structural steel	1.25	1.33	0.15	0.14
Stainless steel	1.26	1.21	0.09	0.10
Aluminium	1.39	1.39	0.04	0.03
Average	1.32	1.38	0.13	0.15

¹ Insufficient experimental data

Table 7: CSM and Eurocode bending resistance prediction comparison for $0.3 < \bar{\lambda}_c \leq 0.6$.

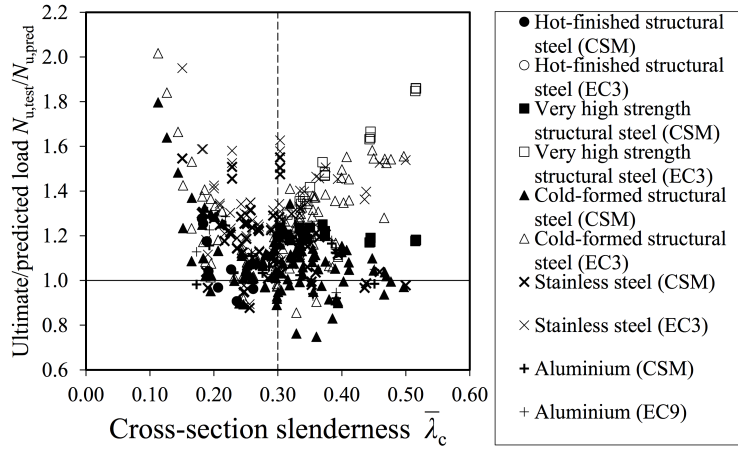


Figure 10: CSM and Eurocode compression resistance prediction comparison.

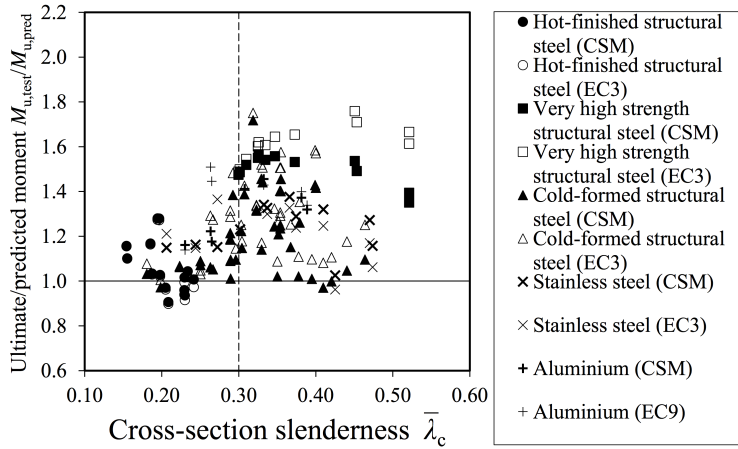


Figure 11: CSM and Eurocode bending resistance prediction comparison.

reach M_{pl} and only approaches it asymptotically. Inspection of Fig. 11 reveals
280 that this increased conservatism is often beneficial as a number of hot-finished
bending results are overpredicted by EN 1993-1-1 [2] and are better predicted
by the lower CSM resistance.

For the slender CHS in compression, the CSM predictions are on average more
285 accurate than those from the Eurocodes for all metallic materials. In terms
of consistency in the compression predictions, there is reduced scatter for very
high strength structural steel and cold-formed structural steel. The COV value
is unchanged for aluminium and slightly higher for stainless steel. Considering
slender CHS in bending, the CSM predictions are the same or more accurate
290 than the Eurocodes for all materials apart for stainless steel. The latter out-
come is due to the EN 1993-1-4 [4] class 3 bending limit for stainless steel being
much higher than those in EN 1993-1-1 [2] and EN 1999-1-1 [5], as apparent
in Fig. 2. This leads to a long $M_{u,pred} = M_{el}$ plateau that extends far beyond
the equivalent structural steel and aluminium class 3 bending limits and the
295 CSM yield slenderness limit, resulting in the CSM bending resistance predic-
tions for slender stainless steel CHS being more conservative than those from
EN 1993-1-4 [4]. This issue is essentially a consequence of comparing a single
harmonised (CSM) design approach against multiple design standards that have
varying class 3 limits in compression and bending and between material types,
300 and also differing reduction factors for local buckling in slender cross-sections.

Overall, the improved capacity predictions of between about 5% and 20% on
average for the different materials, together with the reduced scatter, are clearly
evident in Figs. 10 and 11 and Tables 4 to 7.

305 5. Conclusions

The CSM has been extended to cover the design of CHS and has been seen to
provide improved cross-section resistance predictions of between about 5% and

20% for different metallic materials over traditional design methods. For non-slender CHS ($\bar{\lambda}_c \leq 0.3$) the CSM is generally more accurate and consistent in
310 its capacity predictions and for slender CHS it also offers improved predictions of cross-section resistance, although with more scatter in some cases. The slender results ($\bar{\lambda}_c > 0.3$) are more varied primarily due to the differing, bespoke treatments across the various materials in existing design methods, compared to the standard, harmonised approach across all material types for the CSM.
315 Further work is currently underway in refining the material models, with the aim to gain improved predictions over those currently attained, and undertaking reliability analyses. Improved predictions of CHS cross-section resistances will lead to lighter structures with more efficient material use, leading to more sustainable construction.

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