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## A model of sound scattering by atmospheric turbulence for use in noise mapping calculations

Jens Forssén<sup>1</sup>, Maarten Hornikx<sup>1,2</sup>, Dick Botteldooren<sup>3</sup>, Weigang Wei<sup>3</sup>, Timothy Van Renterghem<sup>3</sup> and Mikael Ögren<sup>4</sup>

<sup>1)</sup>Department of Civil and Environmental Engineering, Division of Applied Acoustics,

Chalmers University of Technology, SE-41296 Göteborg, Sweden.

<sup>2)</sup>Building Physics and Services, Eindhoven University of Technology, Eindhoven, the Netherlands

<sup>3)</sup>Department of Information Technology, Ghent University, Ghent, Belgium

<sup>4)</sup>Department of Environmental and Traffic Analysis, VTI, Gothenburg, Sweden

### Summary

Sound scattering due to atmospheric turbulence limits the noise reduction in shielded areas. An engineering model is presented, aimed to predict the scattered level for general noise mapping purposes including sound propagation between urban canyons. Energy based single scattering for homogeneous and isotropic turbulence following the Kolmogorov model is assumed as a starting point and a saturation based on the von Kármán model is used as a first-order multiple scattering approximation. For a single shielding obstacle the scattering model is used to calculate a large dataset as function of the effective height of the shielding obstacle and its distances to source and receiver. A parameterisation of the dataset is used when calculating the influence of single or double canyons, including standardised air attenuation rates as well as façade absorption and Fresnel weighting of the multiple façade reflections. Assuming a single point source, an averaging over three receiver positions and that each ground reflection causes energy doubling, the final engineering model is formulated as a scattered level for a shielding building without canyon plus a correction term for the effect of a single or a double canyon, assuming a flat rooftop of the shielding building. Input parameters are, in addition to geometry and sound frequency, the strengths of velocity and temperature turbulence.

### 1. Introduction

When acoustic shadow regions appear, creating areas with decreasing sound pressure levels compared with the free field level, sound scattering by turbulence grows in importance. The shadow regions of interest here are those caused by shielding objects such as buildings and other noise barriers. Acoustic shadows caused by upward refraction are similarly affected but are not a focus of the current study. The turbulence of the atmospheric surface layer has previously been shown to increase the noise level behind barriers, mainly at higher sound frequencies (e.g. [1]). In first estimates, the turbulent flow actually caused by a noise barrier itself, has been shown to lead to less significant scattering [2]. Previous studies have shown that models using energy based single scattering approximations are well applicable to the problem (e.g. [3, 4]). Even though a higher precision is expected with wave-

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based models, such as the parabolic equation method (e.g. [5, 6]), the finite-difference time-domain method (e.g. [7]) or the equivalent sources method [8], the single scattering approximation is considered to be accurate enough to serve as a basis for an engineering model; in addition, having large benefits in computational cost. The scattering model developed in [4], based on [3] and on theory known from literature (e.g. [9, 10, 11]), has been used, in simplified forms, in engineering models for noise mapping purpose [12, 13]. In the present paper the aim is to present an engineering model that is more generally applicable, i.e. for built up areas with street canyons and inner yards, in addition to a single screen on ground. Below, we describe the underlying scattering cross section model, the development of a numerically efficient model for non-canyon situations, a parameter study for canyon situations and the suggested engineering model for general urban situations, followed by conclusions.

### 2. Model development

### 2.1. Underlying scattering cross section model

Using the scattering cross section by Tatarskii [9], Daigle [3] created a model for the total scattering into the shadow region created by a noise barrier, as briefly described here for convenience. The scattered intensity, or here rather the mean square acoustic pressure,  $\tilde{p}^2$ , can be written as an integral over a volume V, as

$$\tilde{p}^2 = \int_V \tilde{p}_0^2 \frac{\sigma(\theta)}{r^2} \mathrm{d}V,\tag{1}$$

where  $p_0$  is the incoming, undisturbed pressure from the source,  $\sigma(\theta)$  the scattering cross section as a function of the scattering angle  $\theta$ , and r the distance from the point in volume V to the receiver, where V is defined as all points above the lines of sight from both the source and the receiver to the barrier top (see [3] for further details). Ostashev describes the derivation of the scattering cross section as well as different turbulence models [11]. For the work made here, a homogeneous and isotropic von Kármán turbulence model has been used. Within the inertial range of the turbulence, the scattering cross section is identical to the one for the more simplified Kolmogorov model, which can be written

$$\sigma(\theta) = 0.03k^{1/3} \frac{\cos^2 \theta}{\sin(\theta/2)^{11/3}} \left(\frac{C_v^2}{c_0^2} \cos^2 \frac{\theta}{2} + 0.14 \frac{C_T^2}{T_0^2}\right),\tag{2}$$

where k is the acoustic wave-number  $(k = 2\pi f/c_0, \text{ with } f$  the sound frequency and  $c_0$  the mean sound speed),  $C_v$  and  $C_T$  the structure parameters of velocity and temperature fluctuations, respectively, describing their partial turbulence strengths, and  $T_0$  the mean temperature in Kelvin.

# 2.2. Development of a turbulence scattering model for non-canyon situations

Inherent in the above described modelling is the assumption of a single scattering approximation. In an improved model the incoming pressure,  $\tilde{p}_0$ , in Eq. (1), would be altered due to multiple scattering as well as due to the barrier diffraction. A first order correction for multiple scattering could be to remove the intensity from the incoming field that is estimated to already have been redirected due to scattering by volume elements closer to the source. Here, however, a slightly different approach has been taken, where the scattering is limited by a saturation determined by an assumed smallest value of turbulence strength, as further described below. In addition, for use in a noise mapping model, the scattering should be limited so that the scattered plus diffracted intensity does not exceed that of the open field, i.e. without barrier.

To reduce the numerical cost for evaluating the integral of Eq. (1), the integration is made analytically for constant  $\theta$ -values, i.e. in the azimuthal direction to the source-receiver line, which has been as described previously [14]. Furthermore, since the integrand is a relatively slow-varying function of space, a fine discretization is not needed. Here a grid spacing of 1 m has been used, and the height and length of the integration domain is limited to about the size of the source-receiver distance.

It is evident from Eqs. (1-2) that, if the two terms corresponding to velocity and temperature fluctuations are kept separate, the integrals can be calculated for a given geometry, and the dependence on the factors  $k^{1/3}$ ,  $C_v^2$  and  $C_T^2$  can be inferred later.

The effects of varying the sound frequency and the strengths of velocity and temperature turbulence as well as modelling the air attenuation and the scattering saturation are studied at a later stage. First, the total scattered level is estimated, relative to free field, for a set of geometries and for unit turbulence strengths  $(C_v^2 = 1 \text{ m}^{4/3}/\text{s}^2 \text{ respectively } C_T^2 = 1 \text{ K}^2/\text{m}^{2/3})$ . In the set of geometries, the screen height, h, is varied in M = 20 logarithmic steps from 4 to 80 m. The distances to the screen, from the source,  $d_S$ , as well as from the receiver,  $d_R$ , are each varied in N = 25 logarithmic steps from 10 to 500 m. Thereby a dataset of  $M \times N \times N = 12500$  cases is created (the actual number of calculations is 6500 since only the upper triangle of each  $N \times N$  matrix needs to be calculated, due to symmetry).

For each source-screen distance, a planar fit is made to the scattered level as function of the  $M \times N$  points of varying screen height and screen-receiver distance (in log coordinates). Since a plane can be described by a  $3 \times 1$  vector of coefficients, these vectors are computed and stored for each of the N planes of source–screen distances. Their values are displayed in Tables I and II, for velocity and temperature turbulence, respectively, where the geometric variables have been normalized by  $d_S$ , which turns out to be preferable for later use. When the result for a new geometry is to be calculated, an interpolation between the set of vectors can be made for the wanted source-screen distance, and the resulting  $3 \times 1$  vector of plane coefficients can be used to estimate the scattered level for the screen height and screen-receiver distance of interest. If the source and the receiver are not at the same height, the input geometry to the model is first rotated. (The geometry is shown in Figure 1.) An example estimate of scattered levels were calculated assuming a source-screen distance of  $d_S = 40$  m. The interpolation then uses values at  $d_S = 36.8$  and 43.4 m, which are the two nearest  $d_S$  values used in the precalculation of the data set. The results are compared with those of a direct calculation for  $d_S = 40$  m, as shown in Figures 2 and 3. The maximum errors for these results are less than 3 dB for screen heights varying between 5 and 40 m, and screen–receiver distances varying between 10 and 100 m, for both velocity and temperature turbulence. The mean error is within  $\pm 0.2$  dB and the standard deviation of the error (i.e. the standard error) is about 1 dB. Hence, the model based on this precalculated dataset can be used for calculating the amount of turbulence scatting in non-canyon cases, i.e. with a single obstacle (a building or other noise barrier) and no further reflecting façades.

### 2.3. Parameter study for urban canyon situations

For the canyon situations, flat roofs have been assumed and the default cases have equal roof height. Looking at Figure 4, where the geometric parameters are explained, the default double canyon cases have  $H_S = H_R = H_I$ , whereas for single canyon cases either  $H_S$  or  $H_R$  is zero, and for cases without canyon, both  $H_S$  and  $H_R$  are zero. In the parameter study, the sound frequency and the geometric parameters were varied including three horizontally spaced receiver positions. The variations in canyon geometry are then set up by varying the height of the intermediate building,  $H_I$  (5, 10, 20 or 40 m), the width of the intermediate building,  $W_I$  (0.1, 1, 10, 20, 40 or 200 m), and the width of the street on the source side and on the receiver side,  $W_S$  and  $W_R$  (5, 10, 20, 40, 80, 160, 320 m). The number of parameters, their range of values and other input data are shown in Table III.

Entirely, the set of calculations consisted of 37632 separate cases, including the 8 frequencies. To calculate the scattered level, relative to free field, for each case, the scattering is added energy wise for the different reflection orders. Reflection order zero means that the sound has not been reflected by any facade; reflection order one means one façade reflection, in either source or receiver canyon; etc. The reflections are reduced by assuming an energy absorption coefficient of the façades of  $\alpha = 0.2$ , independent of frequency (as suggested e.g. in ISO 9613-2 for typical buildings). An additional cause for energy reduction at reflection is modelled by a Fresnel number criterion, which reduces the reflections that are sufficiently close to the edge between facade and roof. For this model, the Nord2000 methodology for vertical surfaces has been used [12, Section 5.20, except an adaptation to an energy scattering based model (by using  $10 \log_{10}(S)$  instead of  $20 \log_{10}(S)$ , where S is the effective surface within the Fresnel-zone). The effect of ground is modelled as a doubling of energy both at the source side and at the receiver side. The used receiver height is  $y_R = 1.5$  m and can be seen as an approximation also for the commonly used receiver height of 4 m. In the calculations, reflections up to order m = 15 were used, which, for these settings, was shown by numerical tests to give converging results.

The single scattering approximation leads to overprediction at longer distances, which can be concluded e.g. from the scaling properties described in [4], whereby a saturation of the scattering is modelled here. This is done by multiplying the scattered energy by  $\exp(-2xk^2J_{\rm vonK})$ , where x is the horizontal range of propagation and  $J_{\rm vonK} = 10^{-8}$  m. Here,  $k^2J_{\rm vonK}$  is the total extinction coefficient according to the von Kármán model [11], and the value of  $J_{\rm vonK}$  has been estimated from assuming a rather small outer length scale of  $L_0 = 10$  m and small values of the structure parameters, such that  $C_v^2/c_0^2$  and  $C_T^2/T_0^2$  approximately equals  $10^{-8}$  m<sup>-2/3</sup> by the expression

$$J_{\rm vonK} = \frac{3}{10} \pi^2 A K_0^{-5/3} \left( 4 \frac{C_v^2}{c_0^2} + \frac{C_T^2}{T_0^2} \right), \tag{3}$$

where  $K_0 = 2\pi / L_0$ .

Furthermore, the effect of air attenuation is taken into account, with a level reduction in proportion to the horizontal range, x, using standardized attenuation rates as function of frequency <sup>1</sup>.

When calculating the contribution of each reflection, the scattered level for the corresponding path in the non-canyon situation is first found by interpolation between the pre-calculated set of vectors (from Tables I and II). Then the effects of façade absorption, Fresnel weighting, air attenuation and scattering saturation are included for each individual contribution before they are summed up.

### 2.4. Engineering turbulence scattering model for general urban situations

For the engineering models, the results from the parameter study are first energy averaged over the three horizontally separated receiver positions. One quarter of the calculated cases are for both  $H_S$  and  $H_R$  being zero, i.e. situations without any canyon. It turns out that these 9408 cases, after correcting for air attenuation, are well approximated by a linear fit of geometric variables  $\log_{10} \frac{h}{d_0}$  and  $\log_{10} \frac{h^2}{d_S d_R}$ , in addition to  $\frac{10}{3} \log_{10} \frac{f}{f_0}$ , where reference values  $d_0 = 10$  m and  $f_0 = 1000$  Hz have been used and where the geometrical distances, as depicted in Figure 1, now are interpreted as the effective distances from the source to the mid receiver position over a thin screen. The resulting model for the scattered level, relative to free field, in situations without canyon,  $L_{p, \text{ scat, no canyon}}$ , is written as follows.

$$L_{p, \text{ scat, no canyon}} = b_1 + b_2 \log_{10} \frac{h}{d_0} \tag{4}$$

$$+b_3\log_{10}\left(\frac{h^2}{d_S d_R} + \epsilon\right) + \frac{10}{3}\log_{10}\frac{f}{f_0} - \beta x \quad \mathrm{dB},$$

<sup>&</sup>lt;sup>1</sup> Applying values from ISO 9613, part 1, for standard atmospheric conditions with a relative humidity of 70 %, a temperature of  $20^{\circ}$  C and a static pressure of 101325 Pa.

where the values of  $b_i$  (i = 1, 2, 3) are given in Table IV and where  $\epsilon$  is inserted in order to reduce the scattered level when outside of the domain used in the parameter study, with a value of 0.0012, given by numerical tests. Comparing the engineering model with the detailed results of the parameter study, the standard error is about 2 dB for both velocity and temperature turbulence.

The derived model of the scattered level in the canyon case,  $L_{p, \text{ scat, canyon}}$ , is given as a correction term to the level for the non-canyon case:

$$L_{p, \text{ scat, canyon}} = L_{p, \text{ scat, no canyon}} + \Delta L_{\gamma}.$$
 (5)

The correction term  $\Delta L_{\gamma}$  is estimated as follows (with  $H_0 = 10$  m being a reference height).

$$\Delta L_{\gamma} = \gamma_1 + \gamma_2 \log_{10} \frac{H_I}{H_0},\tag{6}$$

$$\gamma_1 = \begin{cases} 7, & \text{if single canyon} \\ 14, & \text{if double canyon} \end{cases},$$

$$\gamma_2 = \begin{cases} 2H_I/W_S, \text{ if single canyon, on source side} \\ 2H_I/W_R, \text{ if single canyon, on receiver side} \\ 2H_I(1/W_S + 1/W_R), \text{ if double canyon} \end{cases}$$

To estimate the overall accuracy of the final model, the barrier diffraction should be taken into account before calculating the error. Here, a fairy strong velocity turbulence has been used for such an error estimate, with  $C_v^2 = 1 \text{ m}^{4/3}/\text{s}^2$ , together with an assumed road traffic noise source strength of  $L_W = 70, 74, 82, 88,$ 94, 92 and 85 dB(A), for the octave bands 63 Hz to 4 kHz, based on the CNOSSOS-EU model with 5% heavy vehicles and a driving speed of 50 km/h [15]. However, the set of geometries has been slightly limited such that the largest width of the intermediate building,  $W_I = 200$  m, has been omitted. This is due to larger errors appearing for these configurations, which are assumed to be of relatively small overall importance. For the diffraction, the barrier model according Wei at al. [16] is used here, which is comparable to the model for wide barriers by Pierce [17]. The geometrical input used for the diffraction modelling is the same as for the turbulence scattering (also assuming the ground to cause energy doubling) except that only the intermediate building is modelled here, i.e. the diffraction contributions via façade reflections are omitted, whereby the total diffracted level is underestimated. Thereby it is reasonable to assume that the error is not underestimated, considering also the use of a fairy strong turbulence.

Using the above inputs for a single point source and comparing the engineering model with the detailed results of the parameter study, including also a diffraction estimate, the standard deviation of the error in A-weighted level is about 4 dB. Even though further accuracy improvements of the model would be possible, the balance between simplicity and accuracy is deemed appropriate for the purpose of engineering noise map calculation models.

For an intermediate height of  $H_S$  or  $H_R$ , i.e. between 0 and  $H_I$ , it is suggested that a linear interpolation of the level is used. Calculated results (not presented here) have shown that the scattered level is a monotonically increasing function with the height of  $H_S$  or  $H_R$ . The rate of increase is higher closer to  $H_I$ , whereby the linear interpolation corresponds to a conservative estimate in the sense of rather overestimating than underestimating the scattered level. Furthermore, as  $H_S$  or  $H_R$  approaches  $H_I$ , the level converges toward a maximum, whereby results for values of  $H_S$  or  $H_R$  larger than  $H_I$  can be taken as those at  $H_I$ .

Since the model presented here assumes a point source in a domain that varies only in two dimensions, it is suggested that a so-called 2.5D approach is used for sources further down the road, and the width of the intermediate building is taken as the length of the source–receiver line occupied by the building. It could also be stressed that, for use in noise mapping models, the scattering should be limited so that the total level including diffraction does not exceed that predicted for open field, i.e. without barrier.

### 3. Conclusion

A previously established turbulence scattering cross section model for a single noise screen has been used to develop an engineering model for a general urban situation with the possibility to account for a street canyon and an inner yard, assuming homogeneous and isotropic turbulence. As an intermediate step, a numerically efficient model was developed, which was also made to account for multiple facade reflections, and then used for a parameter study. Using the results of the parameter study, the engineering model was developed with the aim to balance computational cost and accuracy. Studying the error for the case without canyons, the engineering model showed an overall standard deviation of about 2 dB in relation to the intermediate model, which in turn showed an error of about 1 dB in relation to the starting model. Hence, by assuming additivity of the variances, the total error can be estimated to have a standard deviation of less than 3 dB. With canyons included, an error estimate for a road traffic noise setting, including also a diffraction contribution, resulted in a standard deviation of the error of about 4 dB(A).

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Figure 1. Geometric set-up for single noise barrier.



Figure 2. Comparison between originally calculated results (grayscale surface) and the best fit plane (black grid) for  $f_0 = 1000$  Hz,  $C_v^2 = 1$  m<sup>4/3</sup>/s<sup>2</sup> and  $C_T^2 = 0$ .



Figure 3. Comparson between originally calculated results (grayscale surface) and the best fit plane (black grid) for f = 1000 Hz,  $C_v^2 = 0$  and  $C_T^2 = 1$  K<sup>2</sup>/m<sup>2/3</sup>.



Figure 4. Geometric set-up for urban canyon situations.

Table I. Values of coefficients to define the planes of scattered levels for a unit strength of velocity turbulence, i.e.  $C_v^2 = 1 \text{ m}^{4/3}/\text{s}^2$  and  $C_T^2 = 0$ , at f = 1000 Hz, for varying values of the source–screen distance,  $d_S$ . The scattered level relative to free field is  $L_{p,\text{scat}} = a_1 + a_2 \log_{10}(d_R/d_S) + a_3 \log_{10}(h/d_S)$  dB, where  $d_R$  is the screen–receiver distance and h is the screen height. For intermediate values of  $d_S$ , interpolation is used.

| $d_S$ | $a_1$ | $a_2$ | $a_3$ |  |
|-------|-------|-------|-------|--|
| [m]   | [dB]  | [dB]  | [dB]  |  |
| 10.0  | -61.6 | 17.9  | -19.5 |  |
| 11.8  | -60.2 | 17.6  | -20.5 |  |
| 13.9  | -59.0 | 17.3  | -21.4 |  |
| 16.3  | -57.9 | 17.1  | -22.4 |  |
| 19.2  | -56.9 | 16.8  | -23.2 |  |
| 22.6  | -56.0 | 16.6  | -23.9 |  |
| 26.6  | -55.3 | 16.4  | -24.5 |  |
| 31.3  | -54.5 | 16.3  | -24.9 |  |
| 36.8  | -53.9 | 16.2  | -25.2 |  |
| 43.4  | -53.2 | 16.2  | -25.3 |  |
| 51.0  | -52.5 | 16.2  | -25.3 |  |
| 60.1  | -51.8 | 16.3  | -25.2 |  |
| 70.7  | -51.0 | 16.4  | -25.0 |  |
| 83.2  | -50.1 | 16.6  | -24.7 |  |
| 98.0  | -49.1 | 16.7  | -24.3 |  |
| 115   | -48.0 | 16.9  | -23.8 |  |
| 136   | -46.8 | 17.1  | -23.3 |  |
| 160   | -45.5 | 17.3  | -22.7 |  |
| 188   | -44.2 | 17.4  | -22.2 |  |
| 221   | -42.9 | 17.6  | -21.7 |  |
| 261   | -41.5 | 17.7  | -21.2 |  |
| 307   | -40.2 | 17.8  | -20.7 |  |
| 361   | -38.9 | 17.9  | -20.3 |  |
| 425   | -37.6 | 17.9  | -19.9 |  |
| 500   | -36.3 | 18.0  | -19.5 |  |

Table II. Same as in Table I except for a unit strength of temperature turbulence, i.e.  $C_T^2 = 1 \text{ K}^2/\text{m}^{2/3}$  and  $C_v^2 = 0$ .

| $d_S$ | $a_1$ | $a_2$ | $a_3$ |  |
|-------|-------|-------|-------|--|
|       | [dB]  | [dB]  | [dB]  |  |
| 10.0  | -58.3 | 14.9  | -11.6 |  |
| 11.8  | -57.1 | 14.5  | -12.0 |  |
| 13.9  | -56.0 | 14.2  | -12.5 |  |
| 16.3  | -54.9 | 13.9  | -13.0 |  |
| 19.2  | -53.8 | 13.6  | -13.5 |  |
| 22.6  | -52.9 | 13.3  | -14.0 |  |
| 26.6  | -52.0 | 13.1  | -14.5 |  |
| 31.3  | -51.2 | 12.9  | -14.9 |  |
| 36.8  | -50.5 | 12.8  | -15.3 |  |
| 43.4  | -49.7 | 12.6  | -15.6 |  |
| 51.0  | -49.1 | 12.6  | -15.8 |  |
| 60.1  | -48.4 | 12.5  | -16.0 |  |
| 70.7  | -47.7 | 12.5  | -16.1 |  |
| 83.2  | -47.0 | 12.5  | -16.1 |  |
| 98.0  | -46.2 | 12.5  | -16.1 |  |
| 115   | -45.4 | 12.6  | -16.0 |  |
| 136   | -44.6 | 12.6  | -15.8 |  |
| 160   | -43.6 | 12.7  | -15.6 |  |
| 188   | -42.7 | 12.8  | -15.4 |  |
| 221   | -41.7 | 12.8  | -15.2 |  |
| 261   | -40.6 | 12.9  | -14.9 |  |
| 307   | -39.6 | 12.9  | -14.7 |  |
| 361   | -38.5 | 13.0  | -14.4 |  |
| 425   | -37.4 | 13.0  | -14.2 |  |
| 500   | -36.4 | 13.0  | -13.9 |  |

| $H_I$      | = | 5        | 10      | 20       | 40  |     |     |     |    | [m]     |
|------------|---|----------|---------|----------|-----|-----|-----|-----|----|---------|
| $H_S, H_R$ | = | 0        | $H_I$   |          |     |     |     |     |    | [m]     |
| $W_I$      | = | .1       | 1       | 10       | 20  | 40  | 200 |     |    | [m]     |
| $W_S, W_R$ | = | 5        | 10      | 20       | 40  | 80  | 160 | 320 |    | [m]     |
| $x_S$      | = | $.5W_S$  |         |          |     |     |     |     |    | [m]     |
| $x_R$      | = | $.05W_R$ | $.5W_R$ | $.95W_R$ |     |     |     |     |    | [m]     |
| $y_S$      | = | .5       |         |          |     |     |     |     |    | [m]     |
| $y_R$      | = | 1.5      |         |          |     |     |     |     |    | [m]     |
| m          | = | 15       |         |          |     |     |     |     |    | [-]     |
| α          | = | .2       |         |          |     |     |     |     |    | [-]     |
| $c_0$      | = | 340      |         |          |     |     |     |     |    | [m/s]   |
| f          | = | 31.5     | 63      | 125      | 250 | 500 | 1k  | 2k  | 4k | [Hz]    |
| β          | = | .023     | .090    | .34      | 1.1 | 2.8 | 5.0 | 9.0 | 23 | [dB/km] |

Table III. Input data to parameter study of turbulence scattering for urban canyon situations. The geometric parameters are explained in Fig. 4. The last five parameters are the maximum reflection order, m, the façade's energy absorption coefficient,  $\alpha$ , the sound speed,  $c_0$ , the octave band centre frequencies, f, and the air attenuation,  $\beta$ .

Table IV. Linear fit coefficients for velocity and temperature turbulence.

|         | Velocity<br>turbulence       | Temperature<br>turbulence    |
|---------|------------------------------|------------------------------|
| $b_1 =$ | $-52.8 + 10 \log_{10} C_v^2$ | $-49.6 + 10 \log_{10} C_T^2$ |
| $b_2 =$ | 11.3                         | 11.5                         |
| $b_3 =$ | -17.1                        | -13.1                        |
|         |                              |                              |