1 **Investigation of Near-Surface Mounted Method for Shear** 2 **Rehabilitation of Reinforced Concrete T-Beams using CFRP Rods**

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7 **ABSTRACT**

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8 This paper presents the results of an investigation of reinforced concrete (RC) T-beams 9 retrofitted in shear with near-surface mounted (NSM) fibre-reinforced polymer (FRP) rods. Six 10 full-scale 4520-mm-long RC T-beams were tested to study the effects of important parameters 11 such as the presence of NSM FRP rods, the presence of steel stirrups, and the steel stirrup ratio. 12 This paper provides an insightful and comprehensive description of the behaviour of 13 strengthened T-beams under increasing load, from the formation of the first crack to ultimate 14 failure. The results of this study and those gathered in the presented database show that existing 15 steel stirrups and strengthening NSM FRP did not diminish each other's effect when failure 16 modes unrelated to shear resistance of RC beams were prevented. The experimental results of 17 this study and those in the database were used to verify a newly proposed model to predict the 18 shear contribution of NSM FRP rods and laminates in RC beams strengthened in shear. The 19 proposed model showed an improved accuracy compared to the existing models in the literature. 20 **Keywords**: Concrete beams; fibre-reinforced polymers; retrofitting; shear resistance; near-21 surface mounted; FRP rods; steel stirrups; design model.

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22 **INTRODUCTION**

23 In recent years, fibre-reinforced polymer (FRP) composites as an innovative strengthening and 24 rehabilitating material have been an attractive topic for practitioners and researchers (e.g., 25 Seracino *et al.* 2007a; Kotynia 2007; Rizzo and De Lorenzis 2007; Galal and Mofidi 2009; 26 Wiwatrojanagul *et al.* 2012; Pellegrino and Vasic 2013; and Mofidi *et al.* 2014). Near-surface 27 mounted (NSM) FRP rods or laminates have provided a promising solution to the problem of 28 shear strengthening (e.g., De Lorenzis and Nanni 2001; Barros and Dias 2006; Kotynia 2007). In 29 this technique, FRP rods or laminates are embedded into pre-cut grooves on the concrete cover in 30 the webs of RC beams. The idea was originated in 1940s when Asplund (1949) grouted NSM 31 steel rods to strengthen a bridge slab in flexure. Fifty years later, Blaschko and Zilch (1999) and 32 Nanni *et al.* (1999) proposed use of NSM FRP rods to rehabilitate RC beams in flexure.

33 Shear strengthening using NSM FRP method was investigated for the first time by De 34 Lorenzis and Nanni (2001) through six RC beams, focusing on variables such as the spacing of 35 the rods, the FRP rod inclination angle, the end-anchorage of the FRP rods, and the presence of 36 internal steel shear reinforcement. In their study, a model was proposed to predict the 37 contribution of NSM FRP rods for the shear resistance of RC beams. Later, the model was 38 updated by Parretti and Nanni (2004) to be applicable to NSM FRP laminates in addition to 39 NSM FRP rods. The model is still the most rational prediction tool proposed for RC beams 40 strengthened with NSM FRP. However, it lacks a state-of-the-art bond model. The researchers 41 assumed a constant value of 6.9 MPa as the average bond stress between FRP and concrete in 42 their design equations. Although the proposed value was deemed practical at the early stages of 43 the investigations on NSM method, further studies showed that the bond stress between NSM 44 FRP and concrete is a function of several parameters including concrete strength and cross45 sectional dimensions and shape of the NSM FRP material (e.g., De Lorenzis 2004 and Seracino 46 *et al.* 2006b). Moreover, in their model the effective bond length of the FRP is limited by the 47 cross-sectional dimensions of the RC beam or the corresponding length to FRP effective strain of 48 4000 µE. However, further studies showed that the effective bond length of the NSM FRP is 49 mainly a function of concrete strength, cross-sectional and mechanical characteristics of the FRP 50 (e.g., Seracino *et al.* 2006).

51 Next group of research studies proposed a constant value for the effective strain of the FRP 52 in their model. Barros and Dias (2006) conducted an investigation on eight rectangular RC 53 beams strengthened with NSM FRP strips. The experimental parameters of their studies included 54 the spacing of the strips, the FRP strips inclination angle, the presence of longitudinal steel 55 reinforcement, and the presence steel stirrups. In their study, a bond stress of 16.1 MPa and an 56 effective strain of 0.0059 were assumed to calculate the shear contribution of FRP. Kotynia 57 (2007) performed eight tests on RC beams retrofitted in shear using NSM laminates. The test 58 parameters included the inclination and spacing of the FRP laminates. The findings suggested 59 that the effective strain in NSM FRP could be assumed as 0.0035 to predict the FRP shear 60 contribution. Anwarul Islam (2009) conducted tests on three RC beams retrofitted in shear with 61 NSM FRP rods to study the effects of NSM FRP rod spacing and the presence of steel stirrups. 62 The results suggested that the NSM effective strain should be taken as one-third of the ultimate 63 strain of the FRP rods in calculating the FRP shear contribution. It is now established that the 64 effective strain of the FRP is dependent on several parameters including the mechanical 65 properties, shape, inclination of the NSM FRP and concrete strength (e.g., Dias and Barros 66 2013). Taking the effective strain of the FRP in the abovementioned studies as a constant value 67 might not lead to accurate predicted results.

68 Several experimental studies were conducted on different test variables without proposing 69 separate design equations to predict the behavior of the strengthened specimens. In the 70 investigation by Rizzo and De Lorenzis (2007), the effectiveness of FRP rods versus FRP strips 71 was evaluated in addition to the effects of various adhesives, FRP reinforcement inclination 72 angle, and spacing between FRP reinforcements. Wiwatrojanagul *et al.* (2012) studied the effect 73 of FRP rod materials (Aramid FRP versus Carbon FRP), the inclination and the spacing of FRP 74 rods on six shear strengthened RC beams with NSM FRP rods. They concluded that decreasing 75 the spacing between FRP rods did not necessarily lead to increases in the shear capacity of the 76 strengthened specimens, since this might lead to concrete cover splitting type of failure mode. 77 Rahal and Rumaih (2011) investigated the effect of FRP rods inclination, FRP rods end-78 anchorage and NSM FRP rods type (FRP versus steel rods) on three 3250 mm-long RC beams. 79 Extending the NSM rods into one specimen's concrete flange increased the shear resistance and 80 prevented the concrete cover splitting in the specimen. Cisneros *et al.* (2012) studied the effect of 81 the inclination, the cross-sectional shape (rods versus strips) and the spacing of the NSM FRP 82 composites on sixteen test specimens. Raj and Surumi (2012) tested ten shear strengthened 83 rectangular RC specimens using NSM FRP rods and strips. The test parameters were the FRP 84 strips/rods inclinations and the spacing between the FRP rods/strips. Jalali *et al.* (2012) 85 investigated the effectiveness of man-made NSM FRP rods in shear retrofit of five rectangular 86 RC beams. The spacing of the FRP rods, the FRP rod angle, and the end-anchorage of the FRP 87 rods were the experimented parameters of their study. Their proposed end-anchorage man-made 88 FRP rods enhanced the shear resistance and the ductility of the test specimens. Note that most of 89 RC beams in these studies had a span length shorter than 4 m (small-scale or medium-scale 90 specimens). Currently, there is a lack of large-scale test specimen in the literature for full-scale 91 RC beams strengthened with NSM FRP. The size effect of RC beams strengthened with NSM 92 FRP can be particularly important. This is due to the fact that the concrete cover thickness plays 93 an utterly vital role in beams strengthened with NSM FRP method. The concrete cover thickness 94 is not proportionately down-scaled compared to the rest of RC beams dimensions for a small- or 95 medium-scale test specimen.

96 Dias and Barros (2008, 2010, 2011, and 2012) conducted comprehensive series of tests to 97 strengthen RC beams with NSM FRP laminates. The test variables in their separate studies 98 included concrete compressive strength, pre-cracking of the specimens, internal steel shear 99 reinforcement, the spacing of the FRP reinforcement, and the FRP reinforcement angle. Based on 100 their experimental results, they proposed a new design model for NSM FRP laminates (Dias and 101 Barros 2013). According to their model, the presence of steel stirrups decreases the effective 102 strain in NSM FRP laminates. The effective strain in the FRP is calculated based on curve fitting 103 of the experimental results of their own test as a function of the sum of the rigidity of transverse 104 steel reinforcement and that of transverse NSM FRP and concrete compressive strength. 105 Although several studies have pointed out that the presence of steel stirrups has a diminishing 106 effect on the shear contribution of FRP for RC beams strengthened with Externally-Bonded (EB) 107 method (e.g., Mofidi and Chaallal 2011; Mofidi *et al.* 2012b; Pellegrino and Vasic 2013; Mofidi 108 and Chaallal 2014a-b), such effect has not been thoroughly investigated for RC beams 109 strengthened with NSM FRP method and its occurrence is in doubt.

110 This need is the main impetus of this study to conduct an extensive, comprehensive, and 111 targeted experimental investigation involving six full-scale 4520-mm-long T-beams with 112 different steel shear reinforcement ratios. The results of this study (and those available in 113 literature) have been used to study the mechanism of shear resistance for RC beams strengthened

114 with NSM FRP and investigate a possible interaction between steel and FRP shear 115 reinforcement. Moreover, a reliable design method have been proposed for shear strengthening 116 of RC beams with NSM FRP rods and laminates that uses a state-of-the-art bond model and can 117 predict possible failure modes. The accuracy of the predicted results have been verified using 118 collected experimental data from the published studies on shear strengthening of RC beams 119 using NSM FRP composites and compared with the results predicted by aforementioned design 120 models proposed by other researchers.

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122 **EXPERIMENTAL PROGRAM**

123 **Description of specimens**

124 The experimental program involved six tests performed on full-scale, 4520-mm-long RC 125 T-beams (Table 1). The T-section had overall dimensions of 508 mm (flange width) by 406 mm 126 (total depth). The thicknesses of the web and flange were 152 and 102 mm respectively (Fig. 1). 127 The control specimens not strengthened with carbon FRP (CFRP) rods were labelled as CON. 128 The specimens labelled as NR were RC T-beams with no internal transverse-steel stirrups. The 129 specimens labelled as MR (moderately reinforced with internal transverse-steel reinforcement) 130 and HR (heavily reinforced with internal transverse-steel reinforcement) had steel stirrups spaced 131 at $s = 3d/4$ for MR and $s = d/2$ for HR, where $d = 350$ mm was the effective depth of the beam 132 cross-section. The specimens in this study are representative of RC elements designed prior to 133 modern codes and therefore the beams in the HR series are qualified as heavily reinforced. In 134 addition, the maximum steel stirrup spacing values proposed by modern codes, *0.5d* in ACI 318- 135 11 or $0.7d_v$ in A23.3-14 (where d_v is the greater of 0.9*d* or 0.72*h*, *h* is the height of the member) 136 are for design and field practice to avoid brittle shear failure. This limit, however, is conservative

137 for shear strengthening research purposes. The specimens strengthened with NSM were labelled 138 as NSM. The spacing between the NSM FRP rods was 130 mm for all the strengthened 139 specimens.

140 The longitudinal steel reinforcement consisted of four 25M bars (diameter 25.2 mm, area 141 500 mm²) laid in two layers at the bottom and six 10M bars (diameter 10.3 mm, area 100 mm²) 142 laid in one layer at the top. The bottom bars were anchored at the support with 90-degree hooks 143 to prevent premature anchorage failure. The steel stirrups (where applicable) were 8 mm in 144 diameter (area 50 mm²). The longitudinal steel reinforcement had a nominal yield strength of 145 470 MPa for the NR and HR series and 650 MPa for the MR series. The steel stirrups had a 146 nominal yield strength of 540 MPa for the HR series and 650 MPa for the MR series.

147 A commercially available concrete delivered to the laboratory by a local supplier was used. 148 The specimens were cast in two phases using separate concrete batches of the same mix design 149 and supplier. NR-CON, HR-CON, NR-NSM, and HR-NSM specimens were cast in the first 150 phase and MR-CON and MR-NSM specimens in the second phase. Standard compression tests 151 on control cylinders yielded an average 28-day concrete compressive strength of 25.0 MPa and 152 29.6 MPa for the first and second phases respectively. For each concrete batch, the scatter among 153 the compression test results for the cylinder specimens was negligible.

154 To apply the NSM FRP shear strengthening system, the following steps were performed: 155 (1) grooves of 15 mm width and 15 mm depth and spaced at 130 mm were made on both lateral 156 sides of the beam; (2) the grooves were cleaned by compressed water and air; (3) two-third of the 157 grooves' volume was filled with epoxy blended according to the supplier's recommendations; (4) 158 a thin layer of the supplier's epoxy adhesive was applied around the 9.5-mm-diameter CFRP

159 rods; (5) the 9.5-mm-diameter CFRP rods were installed into the groove on both sides of the 160 beam; and (6) the excess epoxy adhesive was removed.

161 Sand-coated CFRP rods were used to strengthen the RC T-beams. The tensile strength, 162 elongation at break, and modulus of elasticity of the sand-coated CFRP rods were 1885 MPa, 163 1.27% and 148 GPa respectively. The reported results are the mean values of several test results 164 conducted on the steel bars, concrete specimens and CFRP rods. The mechanical properties of 165 the adhesive as specified by the manufacturer were: 21 MPa bond strength, 1% elongation at 166 break, 75 MPa compressive strength, and 3656 MPa compressive modulus.

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168 **Test setup and instrumentation**

169 All T-beam specimens were tested in a simply supported three-point bending 170 configuration, as shown in Fig. 1. The load was applied at a distance $a = 3d$ from the nearest 171 support, which corresponds to the case of a slender beam in which the shear resistance is 172 governed by the beam action mechanism. The vertical displacement of the beam was measured 173 at the position under the applied load using linear variable differential transducers (LVDT, 150 174 mm measuring range). The longitudinal steel reinforcement was instrumented with a strain gauge 175 at the point of loading. Strain gauges were also installed on the transverse-steel stirrups located 176 in the loading zone along the anticipated plane of shear failure. The deformations experienced by 177 the CFRP rods were measured using strain gauges installed on the rods. These gauges were 178 attached vertically to the CFRP rods before the rods were epoxy-bonded into the pre-cut grooves. 179 The maximum strain range of the strain gauges used in this study was 2% per manufacturer's 180 data.

182 **EXPERIMENTAL RESULTS**

183 **Overall response**

184 Table 2 summarizes the overall experimental results obtained from the tests for all 185 specimens. The results are presented in terms of the loads attained at failure, the experimental 186 shear resistance due to concrete, steel stirrups, and CFRP, and the percent shear capacity gain 187 due to CFRP, defined as $(V_f)/(V_{total} - V_f)$, where V_{total} is the total load at failure. In deriving some 188 of the values in Table 2, the following assumptions were made: (a) the shear resistance due to 189 concrete was the same whether or not the beam was retrofitted in shear with FRP and whether or 190 not the retrofitted beam was reinforced with transverse steel; and (b) the contribution of steel 191 stirrups was the same for both strengthened and unstrengthened beams.

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193 **Crack Pattern and Failure Mode**

194 The load level at which the first diagonal shear crack occurred was of similar magnitude 195 for all specimens. During the loading of control specimen NR-CON (unstrengthened and with no 196 steel stirrups), diagonal shear cracks initiated at the centre of the shear span at a load of 78.8 kN. 197 As the load increased, only one crack widened and propagated until final failure, which occurred 198 at a load level of 122.7 kN. An MR-CON beam (unstrengthened but with steel stirrups spaced at 199 260 mm) experienced the first diagonal shear crack at a load (79.2 kN) similar to that in NR-200 CON, but the final failure occurred at a much higher load (294.0 kN) due to rupture of a steel 201 stirrup. An HR-CON beam (with steel stirrups spaced at 175 mm) showed similar behaviour, 202 with the ultimate load reaching 350.6 kN, also due to rupture of a steel stirrup.

203 Among the NSM-strengthened specimens, beam NR-NSM (no steel stirrups, but retrofitted 204 with 9.5-mm-diameter NSM vertical rods spaced at 130 mm) failed when the load reached 198.0 205 kN. This corresponds to a 61% increase in shear capacity with respect to the control beam, NR-206 CON. Only one diagonal shear crack formed in this beam at an angle of 47° , widening and 207 propagating as the applied load increased. A popping noise was noted throughout the test, 208 revealing the progressive cracking of the specimen. This caused splitting of the concrete cover in 209 which two thick layers of concrete cover (including the NSM FRP rods) split off from the sides 210 of the beam. After losing the NSM rods (no bond stress transfer between FRP and concrete), the 211 beam finally failed in a diagonal tension failure mode.

212 For specimens with steel stirrups and strengthened with NSM (i.e., MR-NSM and HR-213 NSM), parallel diagonal cracks started to open up at relatively equal spacing and at an angle of 214 43° to 47° with respect to the beam axis. Figure 2 shows the spacing between these shear cracks 215 in specimen MR-NSM at failure, where the ultimate load reached 380.0 kN, which is 29.2% 216 greater than the capacity of the corresponding control beam MR-CON. The final failure was in 217 shear due to concrete-cover splitting (Fig. 2). For beam HR-NSM, the ultimate load was 365.0 218 kN, which was 4% greater than the capacity of the corresponding control beam HR-CON before 219 the global flexural failure mode occurred.

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221 **Deflection response**

222 Figure 3 shows the curves representing load versus maximum deflection at the loading 223 point for all specimens. The quasi-linear behaviour of curves corresponding to the control beams 224 and NR-NSM specimens is characteristic of a typical shear failure. It can be seen that the NSM 225 method generally enhanced the overall behaviour of the RC beams. Compared to the 226 unstrengthened control specimens, the strengthened T-beams had higher ultimate loads at failure 227 (61%, 29%, and 4% for NR-NSM, MR-NSM, and HR-NSM respectively) and greater maximum

228 deflections (234%, 4%, and 10% for NR-NSM, MR-NSM, and HR-NSM respectively) compared 229 to their corresponding control specimens. All the tested beams except HR-NSM exhibited a 230 brittle type of behaviour, which was characterized by a sudden drop in the load-displacement 231 curves after the peak load (Fig. 3). In contrast, beam HR-NSM showed a ductile behaviour and 232 failed in flexure.

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234 **Strain response**

235 To obtain a better understanding of the shear resistance mechanism in the strengthened 236 beams, extensive instrumentation was installed for strain monitoring, especially in steel stirrups 237 and NSM rods.

238 *Transverse-steel reinforcement* — Figure 4 presents the response of applied load versus 239 maximum measured strain in the transverse-steel stirrups for the MR and HR series that 240 contained steel stirrups. These behaviour curves indicate that the steel stirrups went through three 241 phases during loading. In the initial phase (phase "A"), no noticeable contribution of the 242 transverse steel to shear resistance was observed. In the second phase (phase "B"), the first 243 diagonal cracks initiated, and the steel stirrups started to contribute to shear resistance. This 244 phase started at an average applied load of approximately 75.0 kN for most specimens (Fig. 4). 245 This is close to the failure load of the unreinforced control specimen (NR-CON) at 78.8 kN and 246 the crack opening load of the control specimen with steel stirrups (MR-CON) at 79.2 kN. 247 Therefore, the ending load of phase "A" (starting load of phase "B") is considered as the 248 cracking load of the concrete specimens in this study. After Phase B (in the third phase), the 249 strain in the steel stirrups continued to increase with increasing load at a greater rate than in the 250 second phase until ultimate failure of the specimen. It was also found that the maximum strain in

251 the steel stirrups that intercepted the principal shear crack became greater than the yielding strain 252 (as labelled in Fig. 4) by a significant margin before the specimen failed.

253 Figure 4 also shows that, under the same applied load, the strain level in the steel stirrups 254 was substantially greater in the unstrengthened control specimens with no CFRP. This implies 255 that the presence of NSM CFRP eased the strain in the transverse-steel stirrups. Moreover, 256 yielding of the steel stirrups occurred earlier in specimens with wider spacing between steel 257 stirrups (MR series) than in beams with tighter spacing between steel stirrups (HR series), 258 although in MR series the yield point of the stirrups was greater than that of HR series (i.e., 650 259 MPa versus 540 MPa).

260 *NSM CFRP rods* — Figure 5 shows the load versus maximum measured strain behaviour in the 261 NSM CFRP rods in the strengthened specimens. The maximum strain was measured in the NSM 262 rods that intercepted the major shear crack on both sides of the RC beams. Note that all the strain 263 values reported in this paper are the maximum measured (captured) values. They are not 264 necessarily the absolute maximum values experienced by the NSM rods (i.e., in cases where the 265 strain gauges did not intercept the principal cracks). The curves in Fig. 5 display a similar 266 tendency to those for steel stirrups, except for the last phase. The CFRP rods did not make much 267 contribution during the initial stage (Phase "A") of loading, but began to stretch at an applied 268 load of 100.0 kN on average for all three series. In the last stage, however, the CFRP strain 269 started to decrease, drastically at times, as the load increased (e.g., HR-NSM-Right, where the 270 curve shows a reversing portion, indicating a strain decrease in the FRP with increasing load). 271 This strain decrease can be related to local concrete cover splitting which reduced the shear 272 transfer between the NSM FRP and the RC beams' concrete core, accompanied by concrete 273 cracking noises in the beams during loading.

274 **Behaviour of Strengthened RC Beams under Increasing Load**

275 The local behaviour of steel stirrups and NSM rods, particularly within the failure zones, is 276 next discussed and analyzed in depth. To enhance the reliability of the results, the measured 277 strain gauge data are further compared with associated crack formation and with patterns that are 278 closely related to one another.

279 *Analysis of Steel Stirrup Behaviour at Failure*

280 Figure 6 shows the measured strain distribution at failure among the steel stirrups (S1-S4) 281 for series MR and HR. The strain curves in solid black bars on the stirrups are drawn to scale. A 282 horizontal line near the bottom of the beam indicates the yield point of the steel stirrups (MR: *ε^y* 283 = 0.0033; HR: ε_y = 0.0027). The inclination of the plane of rupture is assumed to be equal to the 284 principal crack angle with respect to the longitudinal axis as measured at the end of the test. 285 Although the cracking pattern in some specimens was distributed throughout the shear span, the 286 principal shear crack was taken as the crack that passed through the entire cross section of the 287 beams and led to ultimate shear failure. In addition to the strain at ultimate failure in each stirrup, 288 Fig. 6 also shows the applied forces corresponding to the yield point (*Fyield*) of each stirrup at the 289 bottom of each beam (stacked bars). These loads are reported as a percentage of ultimate load 290 (*Fultimate*) reached. The number beside each yielded stirrup indicates its sequence of yielding. 291 Based on Fig. 6, the following observations can be made:

292 1. In general, the stirrups crossing the principal shear crack were highly strained. Yielding 293 of the steel stirrups was observed in all cases.

294 2. For the unstrengthened control specimens (MR-CON and HR-CON), the stirrup located 295 in the critical section yielded first, followed by the other stirrups. This corroborates the 296 observation that the first diagonal crack appeared in the beam web, midway between the support 297 and the load application point. As the load increased, these cracks became wider and progressed 298 simultaneously toward both the support and the load point. For the strengthened specimens (MR-299 NSM and HR-NSM), the first steel stirrup to yield was not located in the plane of ultimate 300 failure. Proper strengthening of the specimens resulted in formation of concrete struts throughout 301 the shear span of the RC beams, leading to a distributed cracking pattern. In this case, the first 302 shear crack that caused the first steel stirrup to yield was not necessarily the principal shear crack 303 that led to ultimate shear failure, e.g., stirrup S4 yielded first in specimen MR-NSM, but the 304 principal shear crack passed through stirrup S3. Similar behaviour was observed in specimen 305 HR-NSM, where stirrup S3 yielded first, but the principal shear crack passed through stirrup S2.

306 3. For the unstrengthened control specimens, yielding of steel stirrups began under a load 307 much less than ultimate load, starting at 69% and 81% of ultimate force for specimens MR-CON 308 and HR-CON respectively. On the other hand, for NSM-strengthened beams, yielding of steel 309 stirrups occurred under higher shear loads, between 94% and 100% of ultimate load. This means 310 that the presence of NSM FRP rods delayed yielding of steel stirrups. However, after stirrup 311 yielding, the residual capacity of the specimens became minimal, if not negligible.

312 4. For specimen MR-CON, the strain in the steel stirrups was not well distributed (only 313 stirrups S2 and S3 that intercepted the principal shear crack were significantly strained). This 314 could have been due to the wider spacing between steel stirrups (260 mm) in this specimen, 315 which allowed the major shear crack to pass through the concrete cross section under a lower 316 applied force than in the HR-CON specimen, which had a stirrup spacing of 175 mm. For the 317 NSM-strengthened specimens (MR-NSM and HR-NSM), strengthening using NSM led to 318 formation of concrete struts throughout the specimens' shear spans. This resulted in a much more

319 even strain distribution among the steel stirrups in the strengthened beams (e.g., MR-NSM) than

320 in the unstrengthened ones (e.g., MR-CON).

321 *Analysis of NSM FRP Rod Behaviour at Failure*

322 The distribution of maximum strain attained by the NSM CFRP rods (F1-F5) on both sides 323 of the RC beams is shown in Fig. 7 for the three NSM-strengthened specimens. The clustered 324 bars on each side of the FRP rods (labelled as L and R) are drawn to scale. The applied forces 325 corresponding to the maximum strain on the FRP (*Fε-max*) are also indicated at the bottom of each 326 beam's figure (stacked bars) in terms of a percentage of ultimate force reached (*Fu*). The 327 following observations were made based on Fig. 7:

328 1. Similarly to the strain in steel stirrups, the NSM FRP rods crossing the principal shear 329 crack were strained the most. The maximum strain in CFRP rods was generally greatest in the 330 middle of the failure zone where the first diagonal cracks initiated and propagated 331 simultaneously towards the support and the compression flange. This non-uniform strain 332 distribution in the CFRP rods over the failure zone shows the importance of the CFRP anchorage 333 length available, which was less for the NSM rods at both ends than for rods in the middle of the 334 failure zone.

335 2. In specimen NR-NSM, although the strengthening NSM FRP proved very effective, the 336 strain in the FRP rods was not well distributed with respect to the shear span. This could have 337 been due to the absence of internal steel stirrups, which prevented proper formation of concrete 338 struts across the shear span. Only NSM rods F3 and F4, which intercepted the principal shear 339 crack, were significantly strained. The rest of the FRP rods did not efficiently contribute to shear 340 resistance. This occurred because the specimen failed in a single-crack pattern which did not 341 intercept FRP rods F1, F2, or F5. These NSM rods reached their maximum strain under a lower

342 load (i.e., they stopped contributing to shear resistance before ultimate failure) compared to the 343 rest of the NSM rods. However, in specimens MR-NSM and HR-NSM, shear strengthening of 344 specimens with NSM FRP resulted in a better strain distribution with respect to the shear span 345 than that in the unreinforced specimen (NR-NSM). This could have been due to formation of 346 concrete struts throughout the specimens' shear spans.

347 3. Some of the NSM rods in the critical zone that were significantly strained reached their 348 maximum strain level under a load less than the maximum load at failure (e.g., F2 in MR-NSM 349 and F3 in HR-NSM). This could have been due to local cracks in the concrete cover or to local 350 debonding of NSM rods that stopped shear transfer between the NSM CFRP rods and the 351 surrounding concrete prior to ultimate failure.

352 *Effect of Steel Shear Reinforcement on the Contribution of NSM FRP to Shear Resistance*

353 One of the test parameters of the current study is this effect of transverse-steel shear 354 reinforcement on the FRP contribution of RC beams strengthened with NSM FRP for shear 355 resistance. For specimens NR-NSM (no transverse reinforcement) and MR-NSM (moderate 356 transverse reinforcement), Table 2 shows that the FRP shear contribution slightly increased 357 instead of decreasing in the presence of steel stirrups (i.e., 56.9 kN for MR-NSM versus 49.8 kN 358 for NR-NSM). In other words, the presence of steel stirrups did not diminish the NSM FRP's 359 shear contribution. Unlike in the EB FRP method, the highly stressed areas around NSM FRP do 360 not significantly overlap/interact the highly stressed areas around the existing steel stirrups. This 361 is mainly due to the fact that the location of the NSM FRP rods is generally taken with distance 362 away from the location of steel stirrups to avoid possible damage to the steel stirrups during 363 groove cutting for NSM FRP. Therefore, the bond quality between the NSM FRP and the 364 concrete is not compromised by the presence of the steel stirrup. Note that in the specimens with

365 steel stirrups (MR-CON, HR-CON, MR-NSM, and HR-NSM), shear failure occurred after the 366 steel stirrups intersecting the principal crack had yielded (i.e., the steel contribution to shear 367 resistance was not affected by the presence of the strengthening FRP). Therefore, similar 368 equations can be used to calculate the shear contribution of steel for both unstrengthened and 369 strengthened specimens.

370 In order to better investigate this effect with more experimental results, a database of more 371 than 69 RC beams strengthened with NSM FRP rods and laminate that failed due to NSM FRP 372 strengthening system failure (debonding or concrete cover splitting) was collected (Table 3). The 373 database includes most of the relevant data in the existing literature of the tested beams 374 strengthened with NSM FRP including the geometric properties of the test specimens and of the 375 FRP composites, the elastic and mechanical properties of the materials used, and the load at 376 failure (V_{total}), and the contribution of the FRP to shear resistance (V_f).

377 Fig. 8 displays the FRP shear contribution for NSM-strengthened RC beams versus the 378 internal steel stirrup reinforcement ratios, $\rho_{sv} = (A_{sv})/(b_w \times s)$, where A_{sv} and *s* are the total area of 379 the cross-section and the spacing of the transverse steel reinforcement. Fig. 8 clearly shows that 380 the shear contribution of FRP did not decrease (in fact slightly increased when considering a 381 linear trend line) with the increase in the shear transverse reinforcement ratio. Furthermore, in 382 order to normalize the shear contribution of FRP for RC beams with different cross-sectional 383 dimensions, the variations of the strengthening factor κ_f versus the increases in ρ_{sv} is represented 384 in Fig. 9. The FRP strengthening factor κ_f is defined as $(V_f)/(V_{max})$, where $V_{max} = 0.25 f_c b_w d_v$ is 385 the maximum shear resistance of the cross-section based on CSA A23.3-14. Fig. 9 reveals that 386 the FRP strengthening factor slightly increases when the steel shear reinforcement ratio 387 increases. It can be concluded that unlike EB FRP method, the presence of the internal steel

388 shear reinforcement does not influence the contribution of FRP to the shear resistance for shear-389 strengthened RC beams with NSM FRP method. These results are in correlation with the 390 experimental results of this study and those reported by De Lorenzis and Nanni (2001) among 391 others.

392

393 **PROPOSED SHEAR DESIGN EQUATIONS**

394 For practical design purposes, a design model is proposed in this paper for RC beams 395 strengthened in shear using NSM FRP rods and laminates, which can predict possible failure 396 modes including debonding of FRP and concrete-cover splitting.

397

398 **Description of model**

399 For RC beams strengthened with NSM FRP rods and laminates, the FRP contribution to 400 shear resistance can be written in the following form (Kotynia 2007, Anwarul Islam 2009, Jalali 401 *et al.* 2012):

$$
V_f = \frac{A_f \cdot E_f \cdot \varepsilon_{fe} \cdot d_{fv}}{s_f} \tag{1}
$$

403 where A_f , E_f , d_{fv} , ε_{ef} , and s_f are respectively the FRP cross-sectional area on both sides of the 404 beam, the FRP rod or laminate modulus of elasticity, the effective shear depth of the cross-405 section, the effective strain in the FRP rod or laminate, and the spacing between the NSM FRP 406 rods or laminates. The effective shear depth can be taken as the greater of 0.72*h* and 0.9*d* as per 407 CSA/S806 (2012). The FRP effective strain is the average of the maximum strain experienced by 408 the actively involved FRP rods or laminates at the ultimate point. Precise predictions of effective 409 strain for all potential failure modes of the specimen constitute an important step towards 410 achieving accuracy in calculating the FRP shear contribution at the ultimate loading stage. The 411 corresponding FRP effective strain at the ultimate point due to the applicable failure mode 412 should be evaluated on each side of the major shear crack. Because failure occurs on the side of 413 the crack with the lesser effective strain corresponding to an applicable failure mode, the value of 414 the lesser effective strain is the governing effective strain and should be used in Eq. (1) to 415 calculate the FRP contribution to shear resistance.

416 *Failure due to Debonding of NSM FRP at the Concrete/Epoxy Interface*

417 Debonding of NSM FRP is a common failure mode for RC beams strengthened using 418 NSM FRP (De Lorenzis and Nanni 2001; Rizzo and De Lorenzis 2007). According to previous 419 experimental pull-off tests on NSM FRP plates, debonding at the concrete/epoxy interface has 420 been the major failure mode among all the failure modes considered (Blaschko 2003; Seracino *et* 421 *al.* 2007a). Failure at the concrete/epoxy interface is characterized by cracking in the concrete 422 layer adjacent to the epoxy-bonded layer. The FRP ultimately debonds from the concrete surface 423 with a thin layer of concrete (in some cases, only mortar with no aggregate attached to the FRP). 424 According to Seracino *et al.* (2007b), the pull-off force (P_{fb}) of the CFRP laminate at the 425 concrete/epoxy interface (failure by cracking in the concrete layer adjacent to the epoxy-bonded 426 layer) can be calculated as follows:

427
$$
P_{fb} = 0.85 \varphi_f^{0.25} \cdot f_c^{10.33} \sqrt{L_{per} E_f A_f}
$$
 (2)

428 where units of Newtons and millimetres are used, φ_f is the debonding-failure plane aspect ratio 429 and is equal to $(d_f)/(b_f)$; b_f is the length of the failure plane parallel to the concrete surface (at the 430 FRP/concrete interface), which for NSM laminates and rods is taken to be the width of the 431 groove plus 2 mm; and *df* is the length of the failure plane perpendicular to the concrete surface, 432 which for NSM plates is taken to be the depth of the groove plus 1 mm, where width and depth 433 of the groove in the current study are taken equal to 1.5 times the rod diameter (based on pull-off 434 tests of Wiwatrojanagul *et al.* 2012). The other important parameter in Eq. (2) is *Lper*, which is 435 the debonding failure plane in cross section and was set equal to $(2d_f + b_f)$ under the assumption 436 that the effective bond length (*Lef*) of the FRP plates or rods was fully available. However, for 437 RC beams strengthened in shear with NSM FRP, the effective length of the FRP might not 438 always be physically available. The effective bond length is the length beyond which any 439 increase in bond length does not translate into an increase in bond force. The pull-off force of the 440 NSM FRP at the concrete/epoxy interface is a linear function of the embedment length of NSM 441 FRP rods on both sides of the major crack, where the effective bond length represents the upper 442 bound of FRP bond length (Seracino *et al.* 2007a). According to Seracino *et al.* (2007b), the 443 effective bond length of the NSM FRP plates is given by:

$$
L_{ef} = \left(\frac{\pi}{2\lambda}\right) \tag{3}
$$

$$
\lambda^2 = \left(\frac{\tau_{\text{max}} L_{\text{per}}}{\delta_{\text{max}} E_f A_f}\right) \tag{4}
$$

446 where

$$
\tau_{\text{max}} = (0.802 + 0.078\varphi_{\text{f}}) f_{\text{c}}^{\prime \text{ 0.6}} \tag{5}
$$

448
$$
\delta_{\text{max}} = \left(\frac{0.976 \varphi_f^{0.526}}{0.802 + 0.078 \varphi_f}\right)
$$
 (6)

449 where units of Newtons and millimetres are used, λ is a constant, and τ_{max} and δ_{max} are 450 respectively the maximum shear stress and the maximum slip, assuming a bilinear bond-slip 451 relationship at the concrete/epoxy interface. The maximum shear stress and maximum slip were 452 calculated on the basis of an empirical equation extracted from a statistical analysis (Seracino *et* 453 *al.* 2007a).

454 As mentioned earlier, in many shear-strengthening field cases, the effective length of the 455 FRP might not always be physically available. For this case, a modification factor (*kef*) is 456 proposed in this study to account for the effect of effective bond length in calculating the 457 corresponding maximum FRP strain for the debonding failure mode, i.e.,

$$
k_{ef} = \left(\frac{k_g L_{total}}{L_{ef}}\right) \tag{7}
$$

459 where *kg* is an experimental coefficient to consider the group effect (the original bond model was 460 calibrated for single NSM FRP bonded to concrete blocks) for all active NSM FRP rods and 461 laminates (*kg* is taken equal to 0.75 and 0.50 for rods and laminates, respectively), *Ltotal* is the 462 total length of the NSM FRP rods or laminates that are actively contributing to shear resistance 463 (i.e., intersecting the principal shear crack with proper bond length on both sides of the crack) at 464 ultimate failure. Therefore, *Ltotal* is the sum of the shorter parts of all NSM FRP rods or laminates 465 that pass the 45 degrees major shear crack (as illustrated in Fig. 10 for a template cross-sectional 466 size and NSM FRP spacing), where the upper bound of the shorter part of NSM FRP rods or 467 laminates is *Lef*. In Fig. 10, *L1* is the dimension of the shorter part of the first NSM FRP that 468 passes the shear crack. *L2* and *L3* are the dimensions of the shorter parts of the second and the 469 third NSM FRP that pass the shear crack, the dimensions of which are limited by the upper 470 bound, i.e., *Lef*. *L4* is the dimension of the shorter part of the last NSM FRP and is represented by 471 term *rf* in the general form calculations below. The total length, *Ltotal*, is indeed a parameter 472 dependent on beam size (*dfv*), NSM FRP spacing (*sf*), and effective length of NSM FRP rods or 473 laminates (*Lef*). *Ltotal* can be calculated for different scenarios as follows:

474 (1) If
$$
\frac{d_{f\nu}}{2} < L_{ef}
$$
:

475
$$
L_{total} = \left(\left[n_f\right] - \left[\frac{n_f}{2}\right]\right) \cdot r_f + s_f \cdot \left[\frac{n_f}{2}\right] \cdot \left(\left[\frac{n_f}{2}\right] + 1\right) = \left(\left[\frac{n_f}{2}\right] + 1\right) \times \left(r_f + s_f\left[\frac{n_f}{2}\right]\right) : \left[n_f\right] \text{ is an odd number (8)}
$$
476

477
$$
L_{\text{total}} = (\left[n_f\right] - \left[\frac{n_f}{2}\right] \cdot r_f + s_f \cdot \left[\frac{n_f}{2}\right]^2 = \left[\frac{n_f}{2}\right] \times (r_f + s_f \left[\frac{n_f}{2}\right]) \qquad \qquad : \left[n_f\right] \text{ is an even number (9)}
$$

479 where
$$
\begin{bmatrix} n_f \end{bmatrix}
$$
 and $\begin{bmatrix} \frac{n_f}{2} \end{bmatrix}$ are the integer parts of $\begin{bmatrix} \frac{d_{fv}}{s_f} \end{bmatrix}$ and $\begin{bmatrix} \frac{d_{fv}}{2s_f} \end{bmatrix}$, respectively, and $r_f = d_{fv} - s_f$

$$
480 \qquad \left[n_f \right].
$$

481 (2) If $\frac{d^2F}{dr^2} \ge L_{\text{eff}}$: 2 *fv ef d* $\geq L$

482
$$
L_{total} = \left(\frac{d_{fv} - 2L_{ef}}{s_f} + 1\right) \cdot L_{ef} + \left[\frac{n'_{f}}{2}\right] \times \left(r'_{f} + s_{f} \cdot \left[\frac{n'_{f}}{2}\right]\right) \tag{10}
$$

483 where $\lfloor n'_f \rfloor$ and $\lfloor \frac{n_f}{2} \rfloor$ are the integer parts of $\lfloor \frac{n_f}{s_f} \rfloor$ and $\lfloor \frac{n_f}{s_f} \rfloor$, respectively, and $\left\lceil n_{\scriptscriptstyle f}^{\prime}\right\rceil$ $\left\lfloor \frac{m}{2} \right\rfloor$ $(2L_{\rm_{ef}})$ $\left(\frac{2L_{ef}}{S_f}\right)$ *f L s* $(L_{\scriptscriptstyle \rm eff})$ $\left(\frac{L_{ef}}{S_f}\right)$ *f L s*

$$
484 \qquad r'_f = 2L_{ef} - s_f \cdot [n'_f].
$$

485 Consequently, the effective FRP strain corresponding to NSM FRP debonding at the 486 concrete/epoxy interface, (**ε***ef-b*), can be written as:

487
$$
\varepsilon_{ef-b} = \frac{0.4k_{ef} \cdot \varphi_f^{0.25} \cdot f_c'^{0.33} \sqrt{L_{per}}}{\sqrt{E_f A_f}}
$$
(11)

488 Note that the equations proposed in this paper are for vertical NSM FRP rods. The 489 corresponding equations for inclined NSM FRP rods and laminates will be presented in a 490 separate study by the authors.

492 *Failure due to Concrete-Cover Splitting*

493 Concrete-cover splitting at the RC beam's web for specimens shear-strengthened with 494 NSM FRP occurs entirely in the concrete when the stresses in the failure zone reach the concrete 495 tensile strength. This failure mode can be expected more often in specimens with weak concrete 496 and tightly spaced NSM FRP. In an experimental study to investigate the bond behaviour of 497 NSM FRP laminates bonded to concrete blocks, Seo *et al.* (2012) proposed a concrete-cover 498 splitting model for grouped NSM FRP bonded to concrete. According to their model, the 499 concrete splitting capacity for NSM FRP laminates bonded to concrete blocks (*Pfc*) can be 500 calculated as follows:

501
$$
P_{fc} = 0.57 \beta \sqrt{f'_c} A_{cf}
$$
 (12)

502 Where f'_c , β , and A_{cf} are respectively the compressive strength of concrete in MPa, an experimental 503 coefficient equal to 0.7, and the surface area of splitting failure of concrete in mm². Seo *et al.* (2012) 504 proposed an equation to calculate the concrete surface area of splitting failure for one NSM FRP 505 laminate as follows:

506
$$
A_{cf} = 2w_f \sqrt{L_{mb}^2 + b_{el}^2} + L_{mb}(t_f + b_{el})
$$
 (13)

507 where w_f , t_f , L_{mb} , and b_{el} , are the width, thickness, embedment length of FRP laminates and b_{el} is 508 taken equal to 0.3*Lmb*. In this study, *Acf* is modified to consider the effect of all active NSM FRP 509 rods or laminates in concrete cover splitting failure for RC beams strengthened in shear with NSM 510 FRP as follows:

$$
A_{cf} = L_{total}(2w_f + 0.3L_{mb} + t_f) \qquad \text{for FRP plates}
$$
 (14)

$$
A_{cf} = L_{total}(3D_f + 0.3L_{mb})
$$
 for FRP rods (15)

513 where the embedment length can be taken equal to minimum of L_{eff} and $\frac{w}{\sigma}$. Hence, the $\left(\frac{d_{fv}}{2\sin\alpha}\right)$ α

514 concrete-cover splitting force can be used to calculate the FRP effective strain corresponding to the 515 concrete-cover splitting failure mode as follows:

516
$$
\varepsilon_{f_{e-c}} = \frac{0.4\sqrt{f_c'}A_{cf} \cdot s_f}{A_f \cdot E_f \cdot d_f}
$$
 (16)

517 Note that the effective strain in the NSM FRP rods or laminates at ultimate load can be 518 taken as the minimum of the strain due to NSM FRP debonding at the split between the 519 concrete/epoxy interface and the concrete cover, i.e.,

$$
\varepsilon_{f_e} = \min(\varepsilon_{f_{e-b}}, \varepsilon_{f_{e-c}}) \tag{17}
$$

521

522 **Verification of the proposed design equations**

523 The experimental contributions of FRP to shear resistance of the retrofitted specimens in 524 this study are compared with the shear resistance predicted by the proposed equations (Table 4). 525 The calculated values of effective strain developed in the NSM FRP rods corresponding to each 526 failure mode are shown in Table 4. The governing failure mode can thus be identified. The FRP 527 shear contribution is then calculated based on the critical effective strain of the governing failure 528 mode. Finally, the experimental results of this study are compared to the results calculated using 529 the proposed equations (Table 4). For specimens NR-NSM and MR-NSM, the governing shear 530 failure mode according to the calculation is concrete-cover splitting, where the calculated FRP 531 shear contribution (*Vf-g*) for these specimens was 47.8 and 52.0 kN respectively. The 532 experimental FRP shear contribution for these two specimens is 49.8 and 56.9 kN respectively, 533 resulting in a calculated V_f to experimental V_f ratios of 0.96 and 0.91 respectively. This indicates

534 a fairly accurate and conservative prediction by the proposed equations compared with 535 experimental results.

536 To further assess the validity of the proposed theoretical predictions, test results from the 537 database are also used (Table 3). The 69 specimens presented in Table 3 all failed due to 538 debonding of NSM FRP or concrete cover splitting (the results of specimens failed with no 539 major contribution of NSM FRP to the shear resistance, those of specimens failed due to 540 unrelated failure modes to the shear resistance, and those of specimens with special anchorage of 541 NSM FRP is not presented here). The accuracy of the predicted contributions of FRP to the shear 542 resistance by the proposed model is compared to that of the existing design equations in 543 literature proposed by De Lorenzis and Nanni (2001), Kotynia (2007), Anwarul Islam (2009), 544 and Dias and Barros (2013). Figures 11(a) to 11(e) show the *Vf cal* values calculated using the 545 proposed model and each of the existing design models versus the experimental values of *Vf exp*. 546 The results of this study show that the proposed model's accuracy $(R^2 = 0.55)$ is superior to the 547 existing design models [Fig. 11(a)]. The proposed model can calculate the shear contribution of 548 FRP for both beams strengthened with rods and laminates with high accuracy ($R^2 = 0.58$ and R^2 549 = 0.53, respectively, when FRP rods and laminates are considered separately). The predicted 550 results by De Lorenzis and Nanni (2001) for NSM FRP rods show reasonable accuracy with the 551 experimental results $(R^2 = 0.34)$ [Fig. 11(b)]. However, the modified version by Parretti and 552 Nanni (2004) to make the model applicable to NSM FRP laminates does not show good 553 correlation with experimental results ($R^2 = 0.15$, the results are not included in Table 3). This 554 could be due to the fact that the model is not originally calibrated for NSM FRP laminates. The 555 results produced by Kotynia (2007) and Anwarul Islam (2009) [Fig. 11(c)] do not show a great 556 correlation with experimental results in database $(R^2 < 0.1)$. Dias and Barros (2013) model

557 (applicable to NSM FRP laminates) predicts *Vf* with reasonable accuracy when compared to 558 experimental results of RC beams strengthened with NSM laminates $(R^2 = 0.44)$ [Fig. 11(d)]. 559 However, when the experimental results directly used by the researchers to calibrate their model 560 are removed, the accuracy of Dias and Barros (2013) model drops significantly $(R^2 \le 0.1)$ [Fig. 561 11(e)]. This could be due to the fact that their model considers the diminishing effect of steel 562 stirrups on NSM FRP contribution to the shear resistance, which the experimental results 563 analysed in this study shows is not as significant as it was observed previously for EB FRP.

564 The predicted results of the models are also compared with respect to the ratio of $(V_{f, cal})/(V_f)$ 565 *exp*). The proposed model and De Lorenzis and Nanni (2001) model produce accurate and 566 conservative values with respect to (*Vf cal*)/(*Vf exp*) ratio (0.89 and 0.90 respectively). Whereas, 567 models by Dias and Barros (2013) and Anwarul Islam (2009) produce non-conservative 568 predictions with (*Vf cal*)/(*Vf exp*) equal to 1.17 and 1.91, respectively.

569 It should be noted that the proposed model is currently the only design model that can 570 predict the failure mode of the shear strengthened beams with NSM FRP. When compared to the 571 experimental results in the database, the proposed model was able to predict the failure mode of 572 the beams with reasonable accuracy, i.e., 84% and 65% correct predicted failure modes for 573 debonding and concrete covered splitting failure modes respectively.

574

575 **CONCLUSIONS**

576 This paper presents the results of an experimental investigation involving six tests on RC 577 T-beams strengthened in shear using near-surface mounted FRP rods. The parameters of the 578 experimental part of this study were: (i) the effectiveness of the NSM FRP rods, (ii) the presence 579 of steel stirrups, and (iii) the steel stirrup ratio. This paper has provided an insightful and broad 580 description of the behaviour of RC beams strengthened in shear with NSM FRP under increasing 581 load. The behaviour of FRP and of steel stirrups has been analyzed in depth, in particular at 582 ultimate stage within the failure zone. The results of this study and those gathered in a database 583 were further used to verify a newly proposed model to predict the shear contribution of RC 584 beams strengthened with NSM FRP. The main findings of this research can be stated as follows:

585 • The near-surface mounting method greatly enhanced the overall behaviour of RC beams 586 because the strengthened beams showed higher load at failure (31% on average) and 587 greater maximum deflection than their corresponding control specimens.

588 x Proper NSM strengthening led to formation of concrete struts throughout the shear span 589 of the RC beams, resulting in a distributed crack pattern. Unlike the control specimens, in 590 the strengthened beams the first shear crack that caused the first steel stirrup to yield was 591 not necessarily the principal shear crack that led to ultimate shear failure.

592 • The experimental results of this study and those presented in the database have revealed 593 that the presence of steel stirrups did not diminish the NSM FRP shear contribution. This 594 might have been due to the fact that the highly stressed areas around the existing steel 595 stirrups and the FRP did not normally overlap in the NSM strengthening method.

596 vielding of steel stirrups were observed in all specimens tested in this study, as generally 597 assumed in the design models for RC beams with steel stirrups strengthened in shear with 598 FRP (e.g., ACI 440.2R-08 and CSA/S806-12).

599 • The proposed design model shows good correlation between the predicted results and the 600 experimental results presented in the database as well as other available design models in 601 literature. The proposed model produces accurate and conservative values with $(V_{f, cal})/(V_f)$ 602 *exp*) equal to 0.89. This new design model is also the only model that can reasonably

603 predict the possible failure modes of strengthened RC beams using NSM FRP rods and 604 laminates.

605

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- 614

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- 706
- 707
- 708
- 709
- 710

Diameter Spacing Spacing (mm) (mm) (mm) (mm) $\overline{}^*$ - 260 8 175 8 130 9.5 NSM rods NSM rods 260 130 9.5 $8\,$ 9.5 130 NSM rods 175 $8\,$ *Not applicable.	Specimen	Strengthening configuration	Internal transverse steel shear reinforcement	NSM FRP rods			
					Diameter		
	NR-CON						
	MR-CON						
	HR-CON						
	${\it NR\text{-}NSM}$						
	MR-NSM						
	$HR-NSM$						

711 Table 1: Description of the test specimens.

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733 Table 2: Experimental results.

Specimen	Load at rupture (kN)	Total shear resistance (kN)	Resistance due to concrete (kN)	Resistance due to steel (kN)	Resistance due to CFRP (kN)	Gain due to CFRP $(\%)$	Deflection at load point	Failure mode [*]
							(mm)	
NR-CON	122.7	81.3	81.3	0.0	0.0	θ	2.6	DTF
MR-CON	294.0	194.7	88.5	106.2	0.0	θ	11.2	DTF
HR-CON	350.6	232.2	81.3	150.9	0.0	θ	11.9	DTF
NR-NSM	198.0	131.1	81.3	0.0	49.8	61	6.1	CCS
MR-NSM	380.0	251.6	88.5	106.2	56.9	29	11.7	CCS/FLX
HR-NSM	365.0	241.7	81.3	150.9	9.5	$\overline{4}$	13.1	FLX/CCS

734 *Note: DTF, CCS and FLX correspond to diagonal tension, concrete cover splitting and flexural failure modes.

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Taone 9. Experimental data of sheaf strengthening								$\frac{1}{2}$		110111111					
Specimen	b_w (mm)	\overline{d}	f'_{c} (mm) (MPa)	FRP shape	ρ_f $(\%)$	ρ_{sv} $(\%)$	E_f (GPa)	\mathcal{E}_{fu}^* $(\times 10^{-3})$	α (°)	V_{f} cal DN (kN)	V_{fcal} КT (kN)	V_{f} cal AN (kN)	V_{fcal} DB (kN)	V_{fcal} Proposed (kN)	$V_{f\,exp}$ (kN)
$DN(B45-7)$	152	356	31	Rods	0.74	0.00	105	18	45	67.9	119.1	177.2	n/a	108.7	75.1
$DN(B90-5)$	152	356	31	Rods	0.73	0.00	105	18	90	32.1	118.1	248.4	n/a	48.7	37.4
DN(B90-7)	152	356	31	Rods	0.52	0.00	105	18	90	11.1	84.2	177.2	n/a	25.6	24.9
$BD(B10-VL)$	150	123	56	Strips	0.19	0.00	150	14	90	n/a	14.6	25.3	47.0	1.22	28.6
$BD(B10-IL)$	150	123	56	Strips	0.18	0.00	150	14	45	n/a	13.8	16.8	89.0	5.4	23.2
$BD(B12-VL)$	150	123	56	Strips	0.37	0.00	150	14	90	n/a	29.3	50.5	59.0	12.6	31.8
$BD(B12-IL)$	150	123	56	Strips	0.35	0.00	150	14	45	n/a	27.6	33.7	82.0	22.2	36.4
$DB(2S-3LV-a)$	180	356	31	Strips	0.06	0.10	167	18	90	n/a	17.6	36.7	28.2	4.8	0.6
$DB(2S-5LV-a)$	180	356	31	Strips	0.10	0.10	167	18	90	n/a	29.4	61.3	41.4	24.0	25.2
$DB(2S-8LV-a)$	180	356	31	Strips	0.16	0.10	167	18	90	n/a	47.1	98.1	56.6	49.6	48.6
$DB(2S-5LI45-a)$	180	356	31	Strips	0.10	0.10	167	18	45	n/a	30.3	44.6	58.4	78.4	41.4
DB(2S-8LI45-a)	180	356	31	Strips	0.16	0.10	167	18	45	n/a	48.2	71.1	71.7	75.6	40.2
$DB(2S-3LI60-a)$	180	356	31	Strips	0.06	0.10	167	18	60	n/a	19.8	30.2	42.1	13.3	35.4
$DB(2S-5LI60-a)$	180	356	31	Strips	0.09	0.10	167	18	60	n/a	33.0	50.3	57.2	38.0	46.2
$DB(2S-7LI60-a)$	180	356	31	Strips	0.13	0.10	167	18	60	n/a	46.3	70.6	67.5	78.2	54.6
RD(NB45-73-a)	200	173	29	Rods	0.97	0.18	146	15	45	70.1	139.1	175.8	n/a	38.5	28.0
RD(NB90-45-b)	200	173	29	Rods	1.12	0.18	146	15	90	48.6	159.6	285.2	n/a	16.0	28.6
RD(NB90-73-b)	200	173	29	Rods	0.69	0.18	146	15	90	30.2	98.4	175.8	n/a	16.0	26.4
RD(NB90-73-a)	200	173	29	Rods	0.69	0.18	146	15	90	30.2	98.4	175.8	n/a	16.0	54.2
RD(NB45-146-a)	200	173	29	Rods	0.48	0.18	146	15	45	35.0	69.6	87.9	n/a	16.7	39.1
RD(NS90-73-a)	200	173	29	Strips	0.44	0.18	122	17	90	n/a	52.2	104.6	53.6	18.6	50.5
RD(NS45-146-a)	200	173	29	Strips	0.31	0.18	122	17	45	n/a	37.0	52.3	41.0	18.6	32.7
JL(VR)	200	214	36	Rods	0.09	0.19	235	15	90	17.3	128.0	248.5	n/a	6.5	33.4
JL(IR)	200	214	36	Rods	0.09	0.19	235	15	45	27.6	120.7	165.7	n/a	19.3	39.9
$DB(3S-10LV-d)$	180	360	59	Strips	0.13	0.10	174	16	90	n/a	41.5	79.7	69.3	65.8	65.9
$DB(3S-5LI45-d)$	180	360	59	Strips	0.08	0.10	174	16	45	n/a	24.3	33.0	83.6	67.6	66.1
DB(3S-5LI45F1-d)	180	360	59	Strips	0.08	0.10	174	16	45	n/a	24.3	33.0	83.6	67.6	85.8
DB(3S-5LI45F2-d)	180	360	59	Strips	0.08	0.10	174	16	45	n/a	24.3	33.0	83.6	67.6	65.4
DB(3S-9LI45-d)	180	360	59	Strips	0.13	0.10	174	16	45	n/a	42.6	57.9	110.1	107.0	101.9
DB(3S-8LI60-d)	180	360	59	Strips	0.11	0.10	174	16	60	n/a	39.9	56.1	100.1	103.3	112.3
DB(5S-5LI45-d)	180	360	59	Strips	0.08	0.16	174	16	45	n/a	24.3	33.0	62.0	67.6	74.9
DB(5S-5LI45F-d)	180	360	59	Strips	0.08	$0.16\,$	174	16	45	n/a	24.3	33.0	62.0	67.6	101.1
$DB(5S-9LI45-d)$	180	360	59	Strips	0.13	0.16	174	16	45	n/a	42.6	57.9	86.8	107	108.9
DB(5S-5LI60-d)	180	360	59	Strips	0.07	$0.16\,$	174	16	60	n/a	26.6	37.4	59.9	60.6	73.4
$DB(5S-5L160F-d)$	180	360	59	Strips	0.07	$0.16\,$	174	16	60	n/a	26.6	37.4	59.9	60.6	72.6

767 Table 3: Experimental data of shear strengthening using NSM FRP.

$DB(5S-8LI60-d)$	180	360	59	Strips	0.11	0.16	174	16	60	n/a	39.9	56.1	77.7	103.4	122.5
$DB(2S-4LV-b)$	180	360	40	Strips	0.08	0.10	171	16	90	n/a	25.8	48.6	41.6	22.0	20.2
$DB(2S-7LV-b)$	180	360	40	Strips	0.13	0.10	171	16	90	n/a	40.7	76.8	57.1	50.6	42.1
$DB(2S-10LV-b)$	180	360	40	Strips	0.18	0.10	171	16	90	n/a	58.0	109.4	71.2	56.3	56.2
$DB(2S-4LI45-b)$	180	360	40	Strips	0.08	0.10	171	16	45	n/a	23.9	31.8	61.2	59.5	53.4
$DB(2S-7LI45-b)$	180	360	40	Strips	0.13	0.10	171	16	45	n/a	41.8	55.7	80.9	107.8	70.7
$DB(2S-10LI45-b)$	180	360	40	Strips	0.19	0.10	171	16	45	n/a	59.6	79.6	92.2	107.8	85.6
$DB(2S-4LI60-b)$	180	360	40	Strips	0.07	0.10	171	16	60	n/a	26.1	36.0	59.5	30.3	49.6
$DB(2S-6L160-b)$	180	360	40	Strips	0.11	0.10	171	16	60	n/a	39.1	54.0	74.3	63.3	54.4
$DB(2S-9LI60-b)$	180	360	40	Strips	0.16	0.10	171	16	60	n/a	58.7	81.0	88.6	98.6	65.3
$DB(2S-7LV-c)$	180	360	19	Strips	0.13	0.10	174	16	90	n/a	41.5	79.7	41.1	31.7	28.3
$DB(2S-4LI45-c)$	180	360	19	Strips	0.08	0.10	174	16	45	n/a	24.3	33.1	35.2	23.3	33.9
$DB(2S-7LI45-c)$	180	360	19	Strips	0.13	0.10	174	16	45	n/a	42.6	57.9	46.4	73.1	48.0
$DB(2S-4LI60-c)$	180	360	19	Strips	0.07	0.10	174	16	60	n/a	26.6	37.4	34.5	19.0	33.1
$DB(2S-6LI60-c)$	180	360	19	Strips	0.11	0.10	174	16	60	n/a	39.9	56.1	43.0	39.7	42.7
$DB(4S-7LV-c)$	180	360	19	Strips	0.13	0.17	174	16	90	n/a	41.5	79.7	34.1	31.7	6.8
$DB(4S-4LI45-c)$	180	360	19	Strips	0.08	0.17	174	16	45	n/a	24.3	33.1	24.0	23.3	26.0
$DB(4S-7LI45-c)$	180	360	19	Strips	0.13	0.17	174	16	45	n/a	42.6	57.9	34.1	73.1	31.7
$DB(4S-4LI60-c)$	180	360	19	Strips	0.07	0.17	174	16	60	n/a	26.6	37.4	23.4	19.0	25.2
$DB(4S-6LI60-c)$	180	360	19	Strips	0.11	0.17	174	16	60	n/a	39.9	56.1	30.7	39.7	35.2
RR(B2)	150	430	44	Rods	0.34	0.19	124	15	90	34.7	76.0	137.0	n/a	37.9	70.0
RR(B4)	150	430	44	Rods	0.47	0.19	124	15	45	72.7	107.5	137.0	n/a	152.1	138.0
$KT(BI-2/3B)$	150	310	39	Strips	0.19	0.19	163	14	45	n/a	40.5	46.5	54.5	23.8	64.6
$KT(BI-3/5A)$	150	310	39	Strips	0.32	0.19	163	14	45	n/a	69.5	79.7	67.8	71.0	100.6
WA(A75-90)	250	234	39	Rods	1.57	0.15	69	16	90	74.1	183.0	336.3	n/a	39.6	45.5
WA(A300-45)	250	234	39	Rods	0.55	0.15	69	16	45	12.4	64.7	84.1	n/a	12.4	43.9
WA(C75-90)	250	234	39	Rods		$0.50 \ 0.15$	$72\,$	13	90	42.0	61.9	244.9	n/a	30.0	47.5
WA(C150-90)	250	234	39	Rods	0.25	0.15	72	13	90	17.0	30.4	122.5	n/a	7.6	43.8
WA(C300-45)	250	234	39	Rods	$0.18\,$	0.15	72	13	45	7.6	21.9	61.2	n/a	7.6	33.7
RS(BR90E)	175	211	35	Rods	0.32	0.00	45	19	90	8.0	15.2	35.8	n/a	8.0	36.8
RS(BR45E)	175	211	35	Rods	0.46	0.00	45	19	45	31.9	21.5	35.8	n/a	31.8	29.4
RS(BS90E)	175	211	35	Strips	0.34	0.00	44	17	90	n/a	15.8	31.7	60.4	10.4	49.0
RS(BS90E100)	175	211	35	Strips	0.34	$0.00\,$	44	17	90	n/a	15.8	31.7	60.4	10.4	61.5
RS(BS90E75)	175	$211\,$	35	Strips	0.46	$0.00\,$	44	17	90	n/a	21.1	42.2	66.3	17.7	39.2

Table 3: Experimental data of shear strengthening using NSM FRP (cont'd).

768 $\frac{1}{\epsilon_{\text{f}u}}$ is the ultimate strain of the FRP.

 769 ** DN = De Lorenzis and Nanni (2001); BD = Barros and Dias (2006); DB= Dias and Barros (2008, 2010, 2011, 2012, 2013);

770 RD = Rizzo and De Lorenzis (2007); AN = Anwarul Islam (2009); JR = Jalali *et al.* (2012); RR= Rahal and Rumaih (2011); 768 $\overline{\epsilon_{f\mu}}$ is the ultimate strain of the FRP.

769 $\overline{\epsilon_{f\mu}}$ is the ultimate strain of the FRP.

770 RD = Rizzo and De Lorenzis (2007); AN = Anwarul Islam (2009); JR = Jalali *et al.* (201

771 KT= Kotynia 2007;

772 Table 4: Comparison of predicted results versus experimental results

Specimen	E_{fe-b}	$\varepsilon_{\text{fe-c}}$	E_{fe-g}	$V_{f\text{-}cal}$ (kN)	$V_{f\text{-flux}}$ (kN)	V_{f-g} (kN)	V_f -exp (kN)	V_{f-g} / $V_{f\text{-}exp}$
NR-NSM	0.0010	0.0009	0.0009	$47.8***$	174.7	47.8	49.8	0.96
MR-NSM	0.0011	0.0010	0.0010	52.0	81.3	52.0	56.9	0.91
HR-NSM	0.0010	0.0009	0.0009	47.8	23.8	23.8	9.5	2.50

^{*} $V_{f\text{-cal}}$, $V_{f\text{-div}}$, $V_{f\text{-exp}}$ are the FRP shear contributions corresponding to the calculated governing effective strain at ultimate load, the calculated values at flexural failure of the beam, and experimental r effective strain at ultimate load, the calculated values at flexural failure of the beam, and 775 experimental results respectively, V_{f-g} is the governing calculated shear contribution of FRP and $\varepsilon_{fe-b}, \varepsilon_{fe-c}, \varepsilon_{fe-g}$ are the corresponding effective strain to debonding and concrete cover splitting and the g 776 *εfe-b*, *εfe-c*, *εfe-g* are the corresponding effective strain to debonding and concrete cover splitting and the governing effective strain to calculate the shear contribution of FRP respectively.

^{**} The shear contribution of FRP corresponding to the governing calculated failure mode is underlined (shear versus flexure failure modes). underlined (shear versus flexure failure modes).

- Fig. 1: Details of the tested specimens: (a) elevation; (b) cross-section of specimens with no steel stirrups; and (c) cross-section of specimens with steel stirrups. Dimensions in mm.
- Fig. 2: Distributed diagonal cracking in specimen MR-NSM.
- Fig. 3: Load versus maximum deflection at the load point.
- Fig. 4: Load versus maximum strain in the steel stirrups: MR and HR series.
- Fig. 5: Load versus maximum strain in NSM FRP rods in each side of the web for all the strengthened specimens.
- Fig. 6: Distribution of ultimate strains in the stirrups within the failure zone
- Fig. 7: Distribution of maximum strains in the CFRP rods.
- Fig. 8: Contribution of FRP to the shear resistance versus steel stirrups reinforcement ratio (the straight line is the linear trend line).
- Fig. 9: FRP strengthening factor $\kappa_f = V_f / V_{max}$ versus steel stirrups reinforcement ratio (the straight line is the linear trend line).
- Fig. 10: Calculation of L_{total} for RC beams strengthened with NSM FRP.
- Fig. 11: Predicted versus experimental result using (a) the proposed model; (b) De Lorenzis and Nanni (2001) for NSM FRP rods; (c) Anwarul Islam (2009); (d) Dias and Barros (2013) for laminates; and (e) Dias and Barros (2013) for laminates (the experimental results that are directly used to calibrate the model are not considered in comparison).