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CCRRECTION OF LASER RANGE TRACKING DATA FOR ATMOSPHERIC REFRACTION AT ELEVATIONS ABOVE 10 DEGREES

J. W. MARINI C. W. MURRAY, JR.

NOVEMBER 1973

GODDARD SPACE FLIGHT CENTER GREENBELT, MARYLAND

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ABSTRACT

A formula for correcting laser measurements of satellite range for the effect of atmospheric refraction is given. The corrections apply above 10° elevation to satellites whose heights exceed 70 km. The meteorological measurements required are the temperature, pressure, and relative humidity of the air at the laser site at the time of satellite pass.

The accuracy of the formule was tested by comparison with corrections obtained by ray-tracing radiosonde profiles. The standard deviation of the difference between the refractive retardation given by the formula and that calculated by ray-tracing was less than about 0.04% of the retardation or about 0.5 cm at 10° elevation, decreasing to 0.04 cm near zenith.

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CORRECTION OF LASER RANGE TRACKING DATA FOR ATMOSPHERIC REFRACTION AT ELEVATIONS ABOVE 10 DEGREES

INTRODUCTION

The correction of tracking data for atmospheric refraction has been exhaustively studied, and many correction formulas have been published [1-6]. For certain earth and ocean physics applications, however, position accuracies of better than a few centimeters are desirable [7], and these accuracies are much greater than required for most previous applications. Out of the work cited, only the approach given by Marini [3], and the expansion and integral evaluations of Saastamoinen [5,6] provide the desired accuracy at lower elevation angles ($10^{\circ} - 20^{\circ}$). In this report Saastamoinen's integral evaluations are incorporated into Marini's continued fraction form to provide relatively simple algorithms for correcting laser range-data using surface meteorological measurements.

REFRACIVITY AT OPTICAL FREQUENCIES

There are a number of formulas [8-11] for the refractive index n of air and for the corresponding refractivity

$$N \equiv 10^6 (n-1)$$
 (1)

all of which have sufficient accuracy for use here. The formula employed is [12]

$$N = \left(287.604 + \frac{1.6288}{\lambda^2} + \frac{0.0136}{\lambda^4}\right) \left(\frac{P}{1013.25}\right) \left(\frac{1}{1+0.003661 t}\right)$$

$$-0.055 \left(\frac{760}{1013.25}\right) \left(\frac{e}{1+0.00366 t}\right)$$
(2)

where

 $\lambda \equiv$ wavelength of radiation in microns

- $P \equiv$ atmospheric pressure in millibars
- $e \equiv$ partial water vapor pressure in millibars
- $t \equiv$ temperature in degrees Celsius

Because air is dispersive at optical frequencies, the group refractivity N_g is also required

$$N_{\rm g} = \frac{d}{df} (fN) = N - \lambda \frac{dN}{d\lambda}$$
(3)

where f is the frequency. The expression for the group refractivity can be written as

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$$N_{g} = 80.343 f(\lambda) \frac{P}{T} - 11.3 \frac{e}{T}$$
(4)

where

P = Total air pressure in millibars
 e = Partial pressure of water vapor (mb)
 T = Temperature (°K)

and

$$f(\lambda) \equiv 0.9650 + \frac{0.0164}{\lambda^2} + \frac{0.000228}{\lambda^4}$$
(5)

which, at the 0.6943 micron wavelength of the ruby laser becomes

$$f(0.6943) = 1.0000 \tag{6}$$

GEOMETRY AND NOTATION

The geometry of the satellite-tracking station configuration is shown in Figure 1. Spherical symmetry is assumed, i.e. the refractivity is taken to be a function of height only. The height h is measured from the tracking station upward. The subscript "0" designates quantities evaluated at the tracking station, the subscript "1", quantities evaluated at the satellite. The ray or phase path between tracking station and satellite is shown as a curved line. A. True range R is the distance along the straight line connecting the tracking station and the satellite, and the true elevation angle E is the angle between this line and the horizontal at the station. The nominal earth radius used is $r_e = 6378 \,\mathrm{km}$, and H is the height of the tracking station above sea level. The latitude of the tracking station is φ degrees above the equator.

EXPANSION FORMULA

The apparent range R_c between the ground station and the satellite as measured by a pulsed system is given by the integral of the group index of refraction along the phase path [13, 14]



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$$R_{e} = \int_{r_{0}}^{r_{1}} \frac{n_{g}}{\sin\theta} dr$$
 (7)

where the angle θ is given by Snell's law for a spherically stratified medium

$$\operatorname{nr} \cos\theta = \operatorname{n}_0 \operatorname{r}_0 \cos\theta_0 \tag{8}$$

The correction sought is the difference between the measured and the true value of the range

$$\Delta R \equiv R_{n} - R \tag{9}$$

The expansion of $\triangle R$ in inverse powers of $\sin\theta_0$, following Marini [3] gives

$$\Delta R \stackrel{\circ}{=} 10^{-6} \int N_g \, dh \cdot \frac{1}{\sin\theta_0}$$

$$- \left[\frac{10^{-6}}{r_0} \int h N_g dh - 10^{-12} N_0 \int N_g dh \right]$$

$$+ 10^{-12} \int (N N_g - \frac{1}{2} N^2) \, dh \left[\cdot \frac{1}{\sin^3\theta_0} \right]$$

$$+ 10^{-12} \int (N N_g - \frac{1}{2} N^2) \, dh \left[\cdot \frac{1}{\sin^3\theta_0} \right]$$

where the range of integration is from the tracking station (h = 0) upward to above the atmosphere $(h = \infty)$. The terms containing the satellite range R that appear in reference [3] can be neglected, as shown in Appendix 1, because (10) is to be applied only where $E > 10^\circ$ and $h_i > 70$ km.

The expansion (10) is not the most useful one for many orbit determination programs because the correction is expressed as a function of arrival angle θ_0 , which may not even be measured, rather than as a function of elevation angle E, which is computed. To convert (10) to the desired for. , the first term of the expansion of the angular correction is used

$$\theta_0 - E \stackrel{\circ}{=} 10^6 N_0 \cot E \tag{11}$$

substituting (11) into (10), and making suitable approximations

$$\Delta R = \left[10^{-6} \int N_g \, dh \right] \cdot \frac{1}{\sin E}$$

$$\left[\frac{10^{-6}}{r_0} \int h N_g dh + 10^{-1/2} \int (N N_g - \frac{1}{2} N^2) \, dh \right] \frac{1}{\sin^3 E}$$
(12)

Equition (12) above is the expansion that provides the basis for the correction formula that is the subject of this report.

EVALUATION OF INTEGRALS

The evaluation of the integrals, appearing in (12), as functions of the pressure, temperature, and relative humidity of the surface air at the tracking station, has been treated by Saastamoinen [6]. For completeness, and because they differ in detail, our evaluations are given in Appendix 2. The results are

$$10^{-6} \int N_g \, dh \stackrel{\circ}{=} \frac{f(\lambda)}{f(\varphi, H)} [0.002357 P_0 + 0.000141 e_0]$$
 (13)

$$\frac{10^{-6}}{r_0} \int hN_g dh = f(\lambda) (1.084 \times 10^8) P_0 T_0 K$$
 (14)

$$10^{12} \int (NN_g - \frac{1}{2}N^2) dh = i(\lambda) (4.734 \times 10^8) \frac{P_0^2}{T_0} \cdot \frac{2}{3 - 1/K}$$
(15)

where

$$f(\varphi, H) = 1 - 0.0026 \cos 2\varphi - 0.00031H$$
 (16)

and

$$K = 1.163 - 0.00968 \cos 2\varphi$$
(17)
- 0.00104 T₀ + 0.00001435P₀

CORRECTION FORMULA

The formula for calculating the range error $\triangle R$ from the satellite elevation E is obtained by approximating (12) by a continued fraction form

$$\Delta R = \frac{f(\lambda)}{f(\varphi,H)} \cdot \frac{A+B}{\sin E + \frac{B/(A+B)}{\sin E + 0.01}}$$
(18)

where

$$A = 0.002357P_0 + 0.000141e_0$$
(19)

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$$\mathbf{B} = (1.084 \times 10^{-8}) P_0 T_0 K + (4.734 \times 10^{-8}) \frac{P_0^2}{T_0} \frac{2}{(3 - 1/K)}$$
(20)

$$K = 1.163 - 0.00968 \cos 2\varphi - 0.00104 T_0 + 0.00001435 P_0$$
(21)

Here

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 $\Delta R = Range correction (meters)$

E = True elevation of satellite

 P_0 = Atmospheric pressure at the laser site (millibars)

 $\tilde{T_0}$ = Atmospheric temperature at the laser site (degrees Kelvin)

= Water vapor pressure at the laser site (millibars)

 $f(\lambda) = 1$ for a ruby laser, and is given by (5) otherwise e

 $f(\varphi, H) = 1$ for a laser site at 45° latitude and at sea level, and is given by (16) for sites at different latitudes φ and elevations H (in km) '

The water vapor pressure e_0 may be calculated from a relative humidity measurement R_h (%)

$$e_0 = \frac{R_h}{100} \times 6.11 \times 10^{-273.15}$$
(22)

In (18) the quantity 0.01 is an empirical constant that serves to compensate for the neglect of higher order terms. The divisor $f(\varphi, H)$ can be factored out of the series (12) and consequently the fraction (18) because the error thereby incurred in the second term of (12) is negligable. The use of the sum A + Bwhere it appears in (18) instead of using A alone is an optional adjustment used to reduce at elevations near 90° a small bias that occurs in the expansion (12) because of approximations made in its derivation.

TEST OF ACCURACY

To test the accuracy of formula (18), which is based on surface measurements, range corrections obtained using the formula were compared with corrections

obtained by ray-tracing radiosonde refractivity profiles. The ray-trace corrections are considered to have state-of-the-art accuracy, so that the differences between these corrections and those calculated from the simpler formulas represent the penalty paid for simplicity in calculation and measurement.

The data used in Figures 2-11 was obtained from the National Climatic Center at Asheville, North Carolina. It consists of radiosonde observations taken near Dulles Airport, Virginia, during the year 1967.

Using the procedure described in Appendices 3 and 4, 634 refractivity profiles were calculated up to a height of 1000 kilometers from the radiosonde observations. The calculated profiles were ray-traced [16] at arrival angles of 10° , 15° , 20° , 40° , and 80° , and the tropospheric errors in range and elevation angle were obtained. The histograms of these errors are shown in Figures 2, 4, 6, 8, and 10. The correction formula (18) was applied using only surface data and the known elevation angle to obtain approximate tropospheric corrections. The differences between these algorithm corrections and the ray-trace corrections were calculated. The histograms of these differences is chown in Figures 3, 5, 7, 9, and 11. The maximum bias of the error remaining after correction was -0.1 cm, and the maximum standard deviation was 0.49 cm at 10° , decreasing to 0.04 cm at 80° .

In addition formula (18) was compared with range corrections obtained by ray tracing (at arrived angles of 10°, 15°, 20°, 40°, and 80°) radiosonde refractivity profiles calculated at Jananarive (85 profiles), Fairbanks, Alaska (200 profiles), Athens, Georgia (200 profiles), Greensboro, North Carolina (200 profiles), and Nashville, Tennessee (135 profiles). The maximum standard deviation of the error in the algorithm at 10° was 1 centimeter and the maximum at 80° was 0.06 centimeters. The maximum mean error of the algorithm at 10° was 0.16 cm and the maximum at 80° was 0.07 cm.

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CONCLUSIONS

An equation that corrects laser range data for atmospheric refraction using surface meteorological measurements has been derived, and a comparison made between the corrections calculated using this equation (equation 18) and the corrections calculated by ray-tracing through a radiosonde profile. The comparison (Figures 2-11) indicates that the differences between the corrections calculated be the two methods are negligible for practical applications. Hence accurate refraction correction of laser range data can be made without the requirement for radiosonde measurements or lengthy ray-tracing algorithms.



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It should be pointed out that only the relative accuracy of the two procedures has been tested, and that errors caused by factors common to both methods are not in evidence. For example, equation (4) for the group refractive index is used both in (18) and in the ray-trace equations, and any error in its magnitude would reflect equally in the corrections. Similarly, the hydrostatic equation used in equation (2-1) and hence (18) is also implicit in the ray-tracing method because the heights that appear in radiosonde profiles are not measured quantities but rather are calculated from the measured pressures, temperatures, and relative humidities using the hydrostatic equation. Also, both methods assume horizontal homogeneity. Saastamoinen [6] has estimated the standard error from such sources to be less than 1 or 2 centimeters at 10° elevation.

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APPENDIX 1

NEGLECT OF SATELLITE RANGE

APPENDIX 1

NEGLECT OF SATELLITE RANGE

The correction

$$\Delta R = \int_{r_0}^{r_1} \frac{n_g}{\sin\theta} dr \cdot R$$
 (1-1)

can be written as

Fi :

$$\Delta R = 10^{-6} \int_{r_0}^{r_1} \frac{N_g dr}{\sin\theta} + \left[\int_{r_0}^{r_1} \frac{1}{\sin\theta} dr \cdot R \right]$$
(1-2)

The expansion of the first term in (1-2), using suitable approximations [3], gives

$$10^{-6} \int_{r_0}^{r_1} \frac{N_g dr}{\sin\theta} = \frac{10^{-6}}{\sin\theta_0} \int_0^{\infty} N_g dh$$

$$-\frac{1}{\sin^3\theta_0} \left[\frac{10^{-6}}{r_0} \int h N_g dh \right]$$

$$-10^{-12} \int N_g (N_0 - N) dh$$

$$+ 2.8.8$$

$$(1-3)$$

The expansion of the bracketed second term in (1-2), which represents the difference ΔR_g between the geometrical lengths of the phase and the straight-line paths between the satellite and the tracking station, can be obtained by expanding equation (A5) of reference [3] in inverse powers of $\sin\theta_0$ giving

$$\Delta R_{,} = \frac{1}{\sin^{3}\theta_{0}} \cdot \frac{1}{2} 10^{-12} \int N^{2} dh$$

$$\cdot \frac{1}{2} 10^{-12} \frac{(f N dh)^{2}}{R} \cdot \frac{\cos^{2}\theta_{0}}{\sin\theta_{0}^{4}}$$
(1-4)

A1-1



The relative error incurred in neglecting the last term in (1-4) is estimated by dividing it by the (dominant) first term in (1-3), ignoring the small difference between the magnitudes of N and N_g

relative error =
$$\frac{1}{2} \frac{10^{-6} \int N \, dh}{R} \cdot \frac{\cos^2 \theta_0}{\sin^3 \theta_0}$$
 (1-5)

The satellite height h_1 is roughly approximated by approximated by R $\sin\theta_0$, and the zenith integral is about 2 meters:

relative error
$$\approx \frac{1}{h_1} \tan^2 \theta_0$$
 (1-6)

where h_1 is the satellite height in meters. Taking $h_1 \ge 70$ km and $\theta_0 \ge 10^\circ$, the error calculated from (1-6) is less than 0.05% which can be neglected.

APPENDIX 2

EVALUATION OF INTEGRALS

From the perfect-gas law, the law of partial pressures, and the hydrostatic equation

$$\frac{dP}{dh} = \frac{Mg(P-e)}{RT} + \frac{Mwge}{RT}$$

$$= \frac{MgP}{RT} - \frac{0.378 Mge}{RT}$$
(2-1)

M = 28.966 = Molecular weight of dry air Mw = 18.016 = Molecular weight of water vapor R = 8314.36 Joules (°K)⁻¹ (Kg - Mole)⁻¹ = Universal gas constant g = acceleration of gravity (m/s) h = height (m)

Combining (2-1) and (4)

$$\int N_{g} dh = -80.343 f(\lambda) \frac{R}{M} \int \frac{1}{g} dP$$
(2-2)
$$+ [30.5 f(\lambda) - 11.3] \int \frac{e}{T} dh$$

The first integral on the right side of (2-2) above can be evaluated using the approximation [15].

$$\varphi = 9.806 [1 - 0.0026 \cos 2\varphi - 0.00031 (H + h)]$$
(2-3)

$$\frac{1}{g} \stackrel{\circ}{=} \frac{1}{9.806} \left[1 + 0.0026 \cos 2\varphi + 0.00031 (H + h) \right]$$
(2-4)

A2-1

from which, integrating the last term by parts,

$$-\int \frac{1}{g} dP = P_0 \frac{1}{9.806} [1 + 0.0026 \cos 2\varphi + 0.00031 (H + \frac{1}{P_0} \int Pdh)]$$

= P_0 / \overline{g} (2-5)

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where \overline{g} is the value of g at the height

$$\overline{h} = \frac{1}{P_0} \int P \, dh \tag{2-6}$$

above the tracking station or H + \overline{h} above sea level. Saastamoinen uses a gravitational constant evaluated at*

$$H + \bar{h} = 7.3 + 0.9 H \text{ km}$$
 (2-7)

From (2-7) and (2-3)

$$\overline{g} = 9.784 (1 - 0.0026 \cos 2\varphi - 0.00028 H)$$

$$\equiv 9.784 f(\varphi, H)$$
(2-8)

where H is the station elevation in kilometers. Saastamoinen has also evaluated the integra!

$$\int \frac{e}{T} dh \stackrel{\circ}{=} \frac{R}{4Mg} e_0 \tag{2-9}$$

where the \overline{g} appearing in (2-9) is set equal to \overline{g} in (2-8) as a convenient approximation. The expression for the zenith integral becomes

$$\int Ndh = 80.343 f(\lambda) \frac{R}{M\bar{g}} P_0$$

+ [30.5 f(\lambda) - 11.3] $\frac{R}{4M\bar{g}} e_0$ (2-10)

^{*}An equivalent result can be obtained by numerically estimating \overline{h} using (2-17) with T₀ set equal to T_e + β H where T_e is the sea level temperature.

$$10^{-6} \int N \, dh = \frac{f(\lambda)}{f(\varphi, H)} \left[0.002357 \, P_0 + \frac{(30.5 - 11.3/f(\lambda))}{19.2} \, 0.000141 \, c_0 \right] \quad (2-11)$$

Neglecting small errors in the second term of (2-11), equation (13) results.

SECOND INTEGRAL

In equation (12), the magnitude of the coefficient of $1/\sin E$ is about 2.4 meters, while the coefficient of $1/\sin^3 E$, is only about $\frac{1}{4}$ centimeters. At $E = 10^\circ$, the magnitude of the first term is about 12 meters, while the second is about half a meter. Consequently the second term need not be as accurately evaluated as the first, and it is sufficient to use the approximation

$$\frac{10^{-6}}{r_0} \int hN_g dh \stackrel{\circ}{=} \frac{10^{-6}}{r_e} \int \frac{80.343 f(\lambda) P}{T} h dh$$
(2-12)

where r_e is a nominal earth radius (6378 km) and the air is assumed to be dry. It is also sufficient to treat g as a constant throughout.

From (2-1), and integrating by parts

$$\int \frac{P}{T} h dh = \frac{R}{Mg} \int P dh$$
 (2-13)

The pressure P in (2-13) is obtained by integrating (2-1)

$$P = P_0 \exp\left[\frac{Mg}{R}\int \frac{1}{T} dh\right]$$
(2-14)

The temperature T is assumed to have a linear slope

$$T = T_0 + \beta h \tag{2-15}$$

and the integration in (2-14) is carried out giving

$$P = P_0 \left(\frac{T}{T_0}\right)^{-Mg/R\beta}$$
(2-16)

A2-3

The integration in (2-13) may now be performed

$$\int P dh = P_0 \cdot \frac{R T_0}{Mg} \cdot \frac{1}{1 - \frac{R\beta}{Mg}}$$
(2-17)

From (2-12), (2-13), and (2-17)

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$$\frac{10^{-6}}{r_0} \int h N_g dh \stackrel{\circ}{=} f(\lambda) \frac{10^{-6} (80.343)R^2}{r_e M^2 g^2} P_0 T_0 K$$

$$= f(\lambda) (1.084 \times 10^{-8}) P_0 T_0 K$$
(2-18)

where g has been set equal to 9.784 and the factor

$$K \equiv \frac{1}{1 - \frac{R\beta}{Mg}}$$
(2-19)

is equal to unity in an isothermal atmosphere ($\beta = 0$) and is equal to about 0.8 in an atmosphere in which the temperature lapse rate is a constant 6°/km ($\beta = -6^{\circ}$ /km).

Rather than use the theoretical value for K given by (2-19), which is based on a constant lapse rate, the value of K used in the corrections equations is taken to be an empirical constant which was determined by solving (2-18) for k and calculating its value by numerically integrating through the atmospheres of the U.S. Standard Atmosphere Supplements, 1966. Using linear regression on the values so obtained, the formula

$$K = 1.163 - 0.00968 \cos 2\omega - 0.00104 T_0 + 0.00001435 P_0$$
 (2-20)

resulted. Here φ is the latitude of the tracking station.

THIRD INTEGRAL

The contribution from the third integral in (12) is only marginally significant, and the term can be approximated by

$$\frac{1}{2} 10^{-12} \int N_g N \, dh = \frac{1}{2} 10^{-12} \, (80.343)^2 \, f(\lambda) \int \frac{P^2}{T^2} \, dh \qquad (2-21)$$

Assuming a constant temperature gradient, and using (2-16)

$$10^{-12} \int \left(N Ng - \frac{1}{2} N^2 \right) dh \stackrel{\circ}{=} \frac{10^{-12}}{4} (80.343)^2 f(\lambda) \frac{R}{Mg} \frac{P_0^2}{T_0} \cdot \frac{1}{1 + \frac{RP}{2mg}}$$
(2-22)

The last factor in (2-22) can be expressed in terms of K using (2-19), giving (15).

APPENDIX 3

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PROGRAM FOR CALCULATING REFRACTIVITY PROFILES FROM RADIOSONDE DATA

APPENDIX 3

PROGRAM FOR CALCULATING REFRACTIVITY PROFILES FROM RADIOSONDE DATA*

RADIOSONDE DATA

Radiosonde observations are measurements of pressure, temperature, and humidity taken from the surface up to the point where the balloon that carries the sensors bursts [1]. The values of temperature, pressure and relative humidity measured at certain standard and significant levels during each balloon ascent from numerous weather stations is available from the National Climatic Center. This data can be used to construct continuous refractivity profiles from the surface up to the point of highest measurement. Above the latter point, the refractivity profile can be extended by assuming a suitable temperature profile.

GEOPOTENTIAL ALTITUDE

The equations used to calculate the refractivity profiles employ the geopotential altitude H [2, p. 217], which is given by

$$H = \frac{1}{G} \int_0^Z g \, dZ \tag{1}$$

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where Z is the geometric altitude, and the lower limit of integration is from sea level (Z = 0). H is in geopotential meters when G equals 9.8 m/sec². The local acceleration of gravity is calculated from the latitude ϕ by [2, p. 488]

$$g_0 = 9.780356 (1 + 0.0052985 \sin^2 \phi - 5.9 \times 10^{-6} \sin^2 2\phi)$$
 (2)

and [2, p. 217]

$$g = \frac{g_0 r_0^2}{(r_0 + Z)^2} (m/sec^2)$$
(3)

Here r_0 is an effective earth radius given by [2, p. 218]

$$r_0 = \frac{2 g_0}{3.085462 \times 10^{-6} + 2.27 \times 10^{-9} \cos 2\phi - 2 \times 10^{-12} \cos 4\phi}$$
 (m) (4)

*This appendix is self-contained. It has separate references, and the notation used differs from that in the rest of the report.

From (1) and (3) the conversion between geopotential and geometric altitude is given by [2, p. 218]

$$H = \frac{g_0}{G} \left(\frac{r_0 Z}{r_0 + Z} \right)$$
 (5)

and

$$2 = \frac{r_0 H}{\frac{g_0 r_0}{G} - H}$$
(6)

VIRTUAL TEMPERATURE

The calculations also make use of the virtual temperature $T_{\rm v}$ [3] which is related to the ordinary temperature T (°K) by

$$T_{v} = \frac{T}{1 - 0.379 \frac{e}{P}}$$
(7)

where e is the partial pressure of the water vapor in the air, and is given by [4, p. 343]

$$e = \left(\frac{R_{h}}{100}\right) (6.11) \ 10^{\frac{7.5 (T-273.15)}{237.3 + (T-273.15)}} (m bar) \tag{8}$$

 R_h being the relative humidity in percent.*

CALCULATION OF GEOPOTENTIAL ALTITUDES

The first step in the calculation of refractivity profiles from the rediosonde measurements of pressure, temperature, and relative humidity is to establish a

^{*}If the dewpoint temperature $T_d({}^{\circ}K)$ is given instead of the relative humidity, e can be calculated from (8) by setting $R_h = 10J$ and $T = T_d$.

table of pressure, temperature, and virtual temperature versus geopotential altitude. The virtual temperatures at the given points are calculated from the measured values of P, T, and R_h using (8) and (7).

To calculate the geopotential altitudes, it is necessary to assume hydrostatic equilibrium [3]

$$d\mathbf{P} = -\rho \mathbf{g} \, d\mathbf{Z} \tag{9}$$

The density ρ is given with sufficient accuracy by [3]

$$\rho = \frac{MP}{RT_v}$$
(10)

The apparent molecular weight of dry air is taken to be [2, p. 289]

$$M = 28.966$$
(11)

and the universal gas constant [2, p. 289]

$$R = 8314.36 \text{ Joules } (^{\circ}\text{K})^{-1} (\text{Kg-mole})^{-1}$$
(12)

Using the assumption that the virtual temperature is a linear function of geopotential height between any two adjacent measured points H_1 and H_2 , (9) may be integrated with the use of (1) and (10) to give

$$\frac{P_2}{P_1} = \left(\frac{T_{v1}}{T_{v2}}\right)^{\frac{GM(H_2 - H_1)}{R(T_{v2} - T_{v1})}}$$
(13)

which may be written as

$$H_{2} = H_{1} + \left(\frac{R T_{v1}}{G M}\right) \frac{\chi \ln (P_{2}/P_{1})}{\ln (1 + \chi)}$$

= $H_{1} + \left(\frac{R T_{v1}}{G M}\right) \ln \left(\frac{P_{2}}{P_{1}}\right) \left(1 - \frac{\chi}{2} + \frac{\chi^{2}}{3} \dots\right)^{-1}$ (14)

A3-3

where

$$\chi = (T_{v2} - T_{v1}) / T_{v1}$$
 (15)

Equation (14) can be used stepwise starting at the known geopotential elevation of the radiosonde station to compute the geopotential altitudes.* In this way the required table of pressure, temperature, and virtual temperature versus height is established.

CALCULATION OF REFRACTIVITY PROFILES

The radio refractivity N is given by the formula^{\dagger} [5, p. 7]

$$N = 77.6 \frac{P}{T} + 3.73 \times 10^5 \frac{e}{T^2}$$
(16)

with P and e expressed in millibars and T in degrees Kelvin.

To calculate N at a given height, i.e., to obtain a point of a refractivity profile, it is necessary to know the values of P, T and e at that height. These are obtained as follows:

The height is converted to a geopotential altitude by adding it to the geometric station elevation to obtain the geometric altitude Z, and applying (5). Using the geopotential altitude so calculated, the temperature and the virtual temperature at the given height are obtained from the table of P, T, and T_v vs. H by linear interpolation. The pressure at the given height is calculated using (13) with P₂, T_{v2} and H₂ replaced by the values associated with the given height. Finally the vapor pressure e is calculated from (7). Substitution into (16) then gives the required refractivity.

†At optical frequencies (2) and (4) of the main text are used.

$$N = 77.6 \frac{P}{T} \left(1 + \frac{7.52 \times 10^{-3}}{\lambda^2} \right)$$

where the wavelength λ is in microns.

^{*}The geopotential altitudes are computed at the radio-sonde stations and are included in the data stored at the National Climatic Center. The altitudes are recomputed both as a check of the self-consistency of the data and also to generate geopotential altitudes consistent with the values of the fundamental constants (R and M, for example) adopted.

A listing of the FORTRANH program with a sample profile calculated from meteorological data taken at Dulles airport on 1 January 1967 is shown in Appendix 4.

Also shown are the surface measurements of temperature, pressure and relative humidity, the tropospheric range error obtained from ray-trace (RANGE ERROR), the tropospheric elevation angle error, the tropospheric range error approximation (RANGE ERROR APPROX) obtained from using equation (18) of the main text, and the difference between the ray-trace and the approximation (RANGE DIFF) for arrival angles of 10°, 15°, 20°, 40° and 80°.

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- 4. Berry, F., Bollay, E., and Beers, N., "Handbook of Meteorology," McGraw-Hill, N.Y., 1945.
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APPENDIX 4

PROGRAM LISTING AND EXAMPLE CALCULATION

	LER UPTICNS - NAME - MAINTIM THOISLINGCHT HAUTSIZER ODOOKS
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-1 SN -0005	COMMEN AT 1500 FOLLOG , TEMPE (100), SELIJM(100), THOUR(100), HGPR(100)
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1 9N- 0005	DIMINGIUN THE MARE (1200) - GMR FP(1200) - JPMREF(1200) - F(1200) -
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-1 5N-0011	
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1 5N 0027	
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1 SN 0035	WH 178 - (6, 947) GG
1 SN 0035	SA7 FORMAT(1X+5HGO = ,015+3)
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.1 SM-0038	
15N 0039	CC = (-2.0D-12) + CC
15N 0041	R = (2+0C0+G0)/80T
1 SN. 0642	WHITE- (0,944) . A
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.1 SN-0044	
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1 SN. 00 16	
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1 SN 0076	M = 1 = 1
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1 SN 0082	DO 1 I = 1.M
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15N 0119	5.10	$1000 \pm 1000 \pm 1000\pm 10000\pm 1000\pm 100$
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-1 5N -0121	213	CONTINUE (CARACISEE)
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ISN 0123		10F - (+60+P) /G) - HSPR(+)
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15N-0120		CHEVED(1) = HEMREF(1) + CONST
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		ALIVE SEA LEVEL GEMREF (I)
	<u>د</u>	CONCRETE: 8 HGPR(1)
16N-0120		D) 766 I = 2.N
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-16N-0130		AUT = G*(R + GMALFP(I))
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10N 0125		$BDT = G \neq (R + HGMREF(I))$
ISN 0133		- GUMREF(13 - 1-0F-34TOP/86T
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	č	SLEK(I) AND SLEKV(I) IN DEGREES NELVIN FRA STORES
ICH		
1 CN 0138		DO 709 I = 1, MM1
101 0130		
TSN 0140		TOP2 = TEMFKV(1+1) - TEMPKV(1)
1 CN 0141		
ISN 0142		$s_{LPK}(1) = T_{OP}1/EOT$
16N 0143		TO SERV(1) = TOP2/DOT
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15N 0146		IF (GFMPFF(1) +LE +HGPH(H) + AND + OF HALL
-16N-0148		10 CONTINUE
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16N- 0161		
ISN 0152		DO 712 T = 1, MT
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	The MTML R. MT Molecular Contraction Contraction
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1 SN 01	161
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SN 0219	SWHEFE(1) = GMREEP(1)Plastroup comme
EN 0220	NPTS = N
	NO #30 K = 1 (N)
SN (221	x(k) = HGMREF(K)
SN CZZZ	DIA IN DEFERACIÓN
SN 0223	CALL ESPINIC (SUM)
ISN 0224	
ISN 0225	
ISN 0226	IF (PRINT)1814,1014,1716
1 SN- 0227-	
ISN 0228	WRITE (6,104)
ISN 0229	
16N 0230	WRITE (6.1P13)
101 0231	
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15N-0234	TELEVILLE CONTRACT AND A CONTRACT AN
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- 1 6N - 0234	
15N 0237	ARITE (0,21)
-15N- 0230	
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15N 0242	DD 23 I = 1+N
1 SN 0243	23 WHITE-(6, 420)HGMRE #(1) +TK(1), HA(1), C(1), HA(1), HA
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ISN 025	1 WRITE (4.3) DIATING OF AN
-1 SN- 425	
ISN 025	3 WRITE (4,10) THETAS
15N-025	A THE AV FURMAT(1X, 19HFLEVATEON ANGLE - TOTOTATION
ISN C25	5 WRITF(6+11) RANGF
1 SN 026	WRITE(4-11) RANGE
TEN 025	7 11 FURMAT(1X, BHRANGE = .115.0, 1X, 10.1K LL JHT IT KD/
101 060	WHITE (6+19) SUM7
101-00E	
	19 FORMAT(1X,22HSUM OF HERACTIVI 7 = 1)19167
I SN VXD	9 19 FUEMAT(1X,22HSUM OF REFRACTIVE 7 = ,515+67
15N 025	9 19 FURMAT(1X,22HSUM OF REFRACTIVI 7 = ,)19.67 WRITE (6.18) GMREFR(1) WRITE (4.18) GMREFR(1)
15N 025 18N 026 15N 026	9 19 FURMAT(1X,22HSUM OF REFRACTIVI 7 = ,115.67 0 WRITE (6.18) GMREFR(1) 1 WRITE (4.18) GMREFP(1) 1 Structure Schnetight = ,015.6.1X,1JHK1_JMETERS)
ISN 020 ISN 020 ISN 020 ISN 020	9 19 FORMAT(1X,22HSUM OF HERACTIVI 7 = ,315.67 0
ISN 020 ISN 020 ISN 020 ISN 020 ISN 020	9 19 FORMAT(1X,22HSUM OF REFRACTIVI 7 = ,315.67 0 WRITE (6.18) GMREFR(1) 1 WRITE (4.18) GMREFP(1) 2
ISN 020 ISN 020 ISN 020 ISN 020 ISN 020 ISN 020	9 19 FORMAT(1X,22HSUM OF REFRACTIVI 7 = ,315.67 0 WRITE (6.18) GMREFR(1) 1 WRITE (4.18) GMREFP(1) 2
ISN 023 ISN 026 ISN 026 ISN 026 ISN 026 ISN 026 ISN 026	9 19 FURMAT(1X,22HSUM OF REFRACTIVI 7 = ,315.67 0 WRITE (6.18) GMREFR(1) 1 WRITE (4.18) GMREFR(1) 2 WRITE (4.18) GMREFP(1) 3 WRITE (6.12) PR(1) 4 WRITE (6.12) PR(1) 5 12 FORMAT(1X,11HFRESSURF = ,015.3,14.2441LL19AFS)

16N 0206	NETHE COLDER TOOLE
15N 0267	WEITE LANTA ANT. METATURE = 1415.J.1540FHOFGREES CELCIUS)
ISN 0268	
15N 0269	
ISN C270	WHITE TING CONTRACTOR FUMILITY # STIDS 7, 1X, 74PERCENT)
ISN 2271	15 FUIDFILLADING DECEMBER 1
ISN 0272	
ISN 0273	ARTIE (4610) 2001 LOVATION ANGLE FRIGE # 1115+2+1X+7HRADIAN3)
15N C274	
ISN (275	
ISN 0276	名氏1 10(17) 第二日本(4) 第二〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇〇
ISN 6277	$FACT7 = C_{+}C_{+}C_{+}C_{+}C_{+}C_{+}C_{+}C_{+}$
ISN 0276	DBF = 1.000 - 2.0026001380 - 0.002320*GMRLEP(1)
15N 0275	BCF = 1.410 - 0.412 - 0.50 - 1.40 - 0.412 - 10 - 2+TK(1) + 0.15 3570 - 41ER(1).
ISN 0280	
ISN CREI	FACT2 = (FACT) FROM (T) = (A205D)
ISN 0232	RACTK - (PACITY CONTY CONTY CONTY
ISN 0263	FACT2 = 1+CUVERVIN
ISN 0284	FAGT2
ISN 0285	FACT3 = 2.(()/FACT3
18N-0286	
ISN 0287	FACTA = FACTA/BCD
15N 0288	Fat = FACT2.4. FACT4
ISN 0289	FAA = (0.002357[C*PR(1)+0.0014150 4411 / 0.0014150
1.SN 0290	FGINS - USINCRETA)
ISN 0291	TOP = FAA + FBB
16N 0292	BOT .=.FSIN1 +_FEG/(TOPALFSINI F.FACT)
15N 0293	REF = TCP/BOT
1 SN 0254	ROIEF = (RUR - ANGERRAI-OUSIAIUS-CHO
ISN 0295	WRITE (6.853) RER
ISN OPEA	-WRITE (4+683) RFR
1 CN 6297	ASS FURMAT(1X, 21HRANGE LIROR APPROX = ,315-5,74 METERS
1 3N 0208	WEITE (64854) RDIFF
134 45 40	WRITE (4,854) RDIFF
100 0877	ASA FORMATILY, 144 RANGE DIFF. = + CIS-6+ 34-CH+
TEN 0301	E50 WRITE (6.17) RNGERR
16N 0302	HRITE (A. 17) RNGCRR
TEN 0303	17 FURMAT(1X+14HRANGE ERROR = +D24+10+1A+10HR1COMETENT
TSN C304	13. COAT INUE
LEN C305	GU TC 604
15N 0306-	6CA_CONTINUE
TSN 0307	OC5 HGPR(1) = HGPR(MP1)
TSN 0306	PHESS(1) = PRESS(MP1)
TEN 0309	TENPC(1) = TENPC(MP1)
ISN 0310	RELAUN(1) = ELHUN(MP1)
15N 0311	TCAY(1) = TCAY(MP1)
15N 0312	THOUR(1) = THOUR(NR1)
15N 0313	TMCNTH(1) = TMONTH(NP1)
ISN 0314	TYEAR(1) = TYEAR(MP1)
15N 0315	1 = 2
15N_0316	IFLINDEX
ISN 0318	53 STOP
16N 0119	END
	APTI - P. OPTI CH3 - NAME - MAINFOPT-01 - LINECHT-UUFSIZED / CH3
COM	SOURCE .F BCDIC . NOLIST . NJDICK OAD .FAFS NUPLET . ID NOANL
1 EN 0003	SUBPEUTINE BRINT (SUM)
177 900E	IMPLICIT REALAB (A-H. U-Z)
12M 0003	GENMEN X(1500)+U(1500)+C(25002++N)75-
134 DUAA	M = NPTS = 1
ISN UUGS	SUN . 0.300
134 4444	$DU \ 1 \ I = 1 \cdot 4$
ISN 0007	2 = 1:009/0L06tu(1+1)/u(1))
15N 0000	1 SUM = SUM + 2+(X(I+1)-X(I))+(((I+1)-J(I))
ISN C009	
19N 0010	FND
ISN OC11	51' YE

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A4-6

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	COMP BLER	CPTICNS - NAMES MAINS OPTHOISLIN CONFERNS SIZE	QUOLK. F.NUFDIT.IC.NOXREF
		SCUPCE DEECCICSNOLLSS SOUTH AND A DOMESTIC	NOF BRANGE AREFRAGE
ISN 000	2	SULT CUTINE BAYTER COMPLETER AND THE PARTY	
15N 600	,	INPUT REALES CALLS AND TOAK ING	
	c	THAYER METHOD FUR RAT DOMETHO	ILE REFRATCI)
	í.	INFUT	UNETLES. THETAG IN
	c	(LIMENEIGNESSE WERSON ANGLE DOWLUTION	FPE IN RADIANS AND
	C	RACTANS, OUTPUT LEVENT IN THE STORE	a company of the second se
	Ç	RANGE CORNER FUN AND ATTIN THE FALL SOLD FOR	IND(1501), RE RATD(1501)
15N 000	4	DIMENSION HOLLINGLAR RELEVANT	
15N 000	5	LIMENSION RELIDELED DEEINGE1901)	
ISN OCO	6	DIMENSION REPORTING REPART	
16N 000	7	THETALL) = THELAU	
ISN OCO	8 4	75 RU 5 I = 1 N	المراجع والمتحد والمتحد والمحادي
15N -000	9		
15N 001	0	5 R(1) = HGT(1) + 6270+000	
15N 001	14	$= 00.61 = 2 \cdot N$	
ISN 001	12	$DELN2 = (1 \cdot CD - G) * RELPRATE I CI + CE CI + $	
15N 00	1.3	-DELR2 = R(1) - R(1-1)	
ISN OC	14	P = C(LN2/RFFINE(1+1))	
15N-00	1.5		
TSN CO	16	TUP = 1.000 - (P/2.000) + (P+2/10.00)	
15N 00	17	AUT = 1.000 - (0/2.000) + (04.2/1000)	
ISN 00	18	6 A(1) = (P+TOP)/(Q+BOT)	
1 SN 00	10	00 7 1. = 2+N	
1 CN 00	20	AA = R(1)/(2.0D(*R(1))	
150 00	21	38 = 2.000+(051N(THETA(1)+C.500) +***	
	22	CC = (R(1) - R(1))/R(1)	
ISN OC	21	DD = (REFRAT(1) - REFRAT(1))/REFINITION	
10M 00	24	$EE = (1 \cdot CD - 6) * DCOS(THETA(1))$	
104 00	25	SINSQ = AA*(BB + CC + CD FEL)	وفيقار الالعنقاف فالتراج والمتحوي فالفا
134.00	26	SINA = (SORT(SINSO)	
154 00		ARGT = DARSIN(SINA)	
15N-00		7 THETA(1) = 2.0DO#ARGT	
ISN CO	20	TAL = 0.000	
ISN O	929		
ISN OG	0.30	AA	
15N 0	G 34	BB = -A(1)/(1.000 + A(1))	
ISN O	032		
ISN G	0.3.8-	A TAL - TAL + CC	
ISN C	034	0HI = 0-000	
I-SN-O	035	$\mathbf{D}\mathbf{O} + \mathbf{I} + \mathbf{I} + 2 + \mathbf{N}$	
ISN O	030	AA = THETALLY - THETAL 1-1)	
ISN O	C 37	$a_{1} = 1.000/(A(1) + 1.000)$	
15N 0	0 38		•
ISN. O	039	th put = put + CC	
ISN Q	740		
16N Q	641		
ISN Q	0042	14 DESTNO(1) = 1.000 + (1.00-6)*REFRAJ(1)	
154 0	043		
ISN C	0044	DU IS THE INGITELIAR (LET) BUCOSI THE FALLET)	
15N - C		DU . DTAN(THETA(1)) - DTAN(THETA(1-1))	
15N (0046	$a_{0} = 1.000 + A(1)$	
ISN (0047		
15N (0048		
LSN (0049	14 RE - BU - (DIN) - R(1))##2+4.00))#R(1)#R()	1) #1 \$1N(7+500 #101) ##6
TSN	00 50	RANGSU - DEORT (RANGSU)	
ISN	0051	HABUE - DE - RANGE	
1 SN	C052	NDIC HH * HILL CONTRACT	
		TOD - DEOSTTALL - CSIN(TAU)*CTAN(THITAL)	
13N	0093	TOL = TOP - (REFIND(N)/REFIND(1))	
15N	C054	TOT . LOTE INDEN JAREF ING (1) + ATANL THETAL	* * *
15N	0055	DIT - DOT - DSTN(TAU)	
15N	0056	DUT A DUT A DOUGITAUSMETANETHETAENET	
15N	6097	HUT A DUT OF LOGARCES	
TSN	0055	LPS S CALARYIMPYORTY	
19N	00.44	MALE AND A CONTRACT OF A CONTR	
1514	00.60	END State 2 Mar	

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// DC B= (RE CFM=CBC=M+DULLES-// DC B= (RE CFM=FE-LRE CL=36,BLKSIZE=2160,DEN=3). // VOL=SEF=68CP10 // VOL=SEF=68CP10

4-03-84740-03/4

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-Print - -1.60 - #Avelength-in Hicrons - 3.694300000-39 - Uptical Freqs - 1.00 0.17453293 RADIANS ANGLE = C. 16175540- RADIANS-----------ANGLE---0.34906586 FADIANS ANGLE = ----- G.49813172 - PADIANS -------ANGLE RACIANS 1.37626344 ANGLE = ----GD = 0.98015955D 01 _.R = ... 0.63826800D 07 ---------- -----------FGT GMP PRESS(ME) TEMP PEGC RELH PCT CALHGT ------..... -55+0 -----96.0 108.676 -3.3 1060.0 109.0 -144-508 ---------170.0 100.0 432.566 \$60.0 -2.5 440.0 - 516+966-140.0 -980.0 520.0 CO2.27C 3.8 94.0 940.0 610.0 4.4 -411-0 \$56.084 3.0 80.0 900.0 957.0 45.4 1397.716 -0.4 1400-0 1416.553 78.0 1418.0 850.0 0.2 1560-190-1550-0 836+0 1906.755 2.2 40.0 800.0 1908.0 2426.045 --41-0 -7 60 .0-2420.0 2697.377 43.0 -0.7 2690.0 725.0 2979.0 -7.00+0 3557.612 650.0 -8.3 46.0 3550.0 4033-814 - 55-0. 411.0. 0.0434 4159.741 99.0 -12.9 601.0 4160.0 .99+0 -417-2-460 ----. -13-0 4178.0 600.0 4831.822 99.0 -16.5 550.0 4820.0 91.0 6310.109 450.0 -27.4 6300+0 -6536+829-436-0 -20-1 -- 6520-0 7150.008 47.0 400.0 -31.3 7154.0 .. 380+0 -36.7-8689.0 -42.4 0.0 G140.177 300.0 9142.0 .0.0 .9724.547 9730-0 0.0 10353.468 -50.4 10355.0 250.0 0.0 11774.728 0.0 11931.751 11540.0 -61.9 155.0 ---- 0.0 12695.730---178.0 0.0 13550.018 150.0 -62.1 13550.0 145+0 - -.60..7 0.0 15255.727 -60.0 114.0 15270.0 -63.2. 100.0 16045-0 0.0 16446.634 -64.8 94.0 16440+0 0.0.17433.840 -.-63.5 20.00. 17433-0 0.0 18255.949 -62.4 18254.0 70.0 0.0.19207.971 19211-0 60.0 0.0 20345.012 -59.9 50.0 20340.0 0.0 21740.404 -\$1742.0 40.00 0.0 23541.906 23543.0 30.0 -59-3 0.0 24684.161 -59.2 25..0. 0.0 26052.822 -59.1 20.0 26084.0 0.0 27887.684 -56.8 27389.0 ۵مقد 0.0 30436.001 -58.2 10.0 30435.0 0.0 32692.329 -56.3 32697.0 7.0 0.0 33672.990 -55.6 33677.0 6.0

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				вн(PCT)	R3F 4E1	FEFCRY	REFRAT	GR REFRAT
H(KM)	TENPLK)	pr(MB)						
						<u>- 6-289484560-03</u>		
	92.944	100-00	-+			2 0.285u94960 03	0.2P8560010 03	A 90586790 03-
0.050	270.8	996.67	4.9490	62.06			-0*565113560-03	0.20095520C 03
			-9616+3-		0.963618530 0	12 0.261 347860 J3	0.284153357 0	
151.0	271.5	584.17	5.2663			69 963399663*8 *		0.267792970 03
	6. 113				0.550077890 J	12 9.27826025D 03	0.251050590-03	-0-04505965-0-
6.252	271.1	971.75	5.1724		- 01150195910-	12-0-276721760-03	A STROKED 03	0.28450501D 03
		565=57	003103	00-00-	0.257943710	12 0.275CB150D 03	C. 02515650 03	0.201702410-03
ESE.O	270.7	14.645	5.1167		011-92-20-0-	2-0-272464140-0 3		0.27809656C 03
				53.21	0.305124830	02 0.26893010U V3	50-002500 0	-0-00000000000000000000000000000000000
0.455	273.4	547.21		61036	-495169E5E*0-	97- 88-969-96 -76	0.26452820D 03	0.270890230 03
+-507-			5990 ° 6	63° 62	0.349602030	75 0*5051015100 -	-0-562657170-03	-0-569014870-83
6.558	277.1	500.00	1.2500	4C+43	-05594526E*0-	0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-	0.2(087541D 03	3 0.26714901D 03
- ororo		05626	6.5875	79.18	0.316164830	0.001000000000000000000000000000000000	- 0-56360400-03	9-0-265597430-00
0.661			-6-3-6-7-	****	-0-3000001320		0 .25807614C 0	
	276.9	911.86	6.2774	78.29	100/553000	9-0-1110-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0		<u>3 0.262900000 03</u> 3 0.26165610D 03
		-99646-		25.864	CIERADEAL A	05 0.25303950D 0	0 0.25551125550 8	0.240354980-03-
0.369	276.2	900-25	6.0686	29•90	-0-2061050	32 0 - 251 779520 - 0		3 0.25905971C 03
120.0	- Brokenn				0.263630830	32 0.7565219dh 0	0 021211130 0	3-0-257763760-03-
0.973	275.4	288•7C	3120°C	Ú-18	-0-278A57210-		1 0 25044699F 0	3 0*2564691 ND 03
	336.0		5,5759	£2.4	5 0.273285870	07 (*248009500 0 2 0 246009500 0	3-0-34536436D-0	3-0-565376060-04-
1.079	274.0			9-63		1, 0,24550326D 0	3 0.24792264D 0	3 0.253883720 03
	7. E72	865.76	5.3236	6.5	0.26202540	12 0 34426189D 0	3 0,246662455-0	-0 05130413D 0-
				54.7	7 0.252684220	02 0.243091960 0	0 05251570 0	-0-340018740-03-
1.290	272.9	854•37 2 0 1 0 1	2007 C		-649102055-0-0		3 0 24054357D 0	3 0.246326250 03
			4.004	1 53.1	• 0.19591624D	J2 0.535555550 10	0 044740755.0 E	
16E.I			HAD CO			12 0_23361888D (3 0.2360C673F G	0 000000000000000000000000000000000000
	276.4	332-01	3.026	0°5E (8 0.1462/3320 0 0 14515195	13 0 -232220130 -C	- 3605345540 5	0.238808600 03
	Carde	3-10-1	900.0		011-956-110	32 0.230 E4317D	COCCOSCOSCOSCO E	50 011622220 CO
1.61	276.1	821.01	019.5		9+2+282+1+0-5	-19-10-530461660-1	12 0.230415290	03 0.235953570 63
	6-9-C		2.915	L-5E 0	0 0,141678533	- 05/200220-0	-39023082-0 EC	03 0.434531000 03
1.72	273.1		583 S			00000000000	03 C.23764523D (ED GOTTOTIO CO
		55.207	2.559	2 40.0	11636361+0 10	0. 0000000 0 50	03-0-226262490	0 095055555 00
		33-262	2.624			JZ 0.2226446D	03 0,224923365	03 0.225061470-03
	274.9	768.48	2.789	40.	recented to the second se	- 050505 155 0 - FC	03 0.233567840	03 0.227557060 03
			1911-5		0.133382630	02 0-219563200	01 0 • 227215720 0 50	01 0126126126370 03
0.4	9 274 5	177.76	2.719		13100121-0-55		03 0. 01 05 0 30 30 20	03 0.224799390 03
9					66 0.13043495	0 22 0.217 290310		03-0+1924523+0-0-
2.16	0 274.4	767.16	2002		76-0-12897063	2.02-0.21556628U	N3 0.21694466C	03 0.22205662D 03
15-6	1-676			0	87 0.127512635	0 07 0.214643690	0.1.0.215516550-	03 0-220690630
2.27	1 273-6	5 756•CI	2.5		97-0-12666394	057952110-0-54-0	03 0.21416286C	03 0.219310350 0:
100		746-14	0 E	19 41.	30 0.12546743			
2.38	13 Z7.4*							

REFRAT

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	0-121	55-342-			CLEAT2221-0-	7	U.210652345_03_0.212812812A3C_03_0.217927490 V	1 2
204	272.6	735.74	2.5161	42.12	0.1246795JD	3	0.269319550 03 0.211400040 V3 V.214174120 01	
	275-2	230-52		42-54	-0-124533AJD	3	0.207591570 03 0.210125000 03 0.1307560 01	2
	275.5	725-42	2.4572	42.97	0.124285850	ð	0.20666721D 03 0.20875542D U3 U.4136V3000 V3	
	- 222-0	- 220-53		A2+23	0-121276723	à	0.2055540.40.03 0.207002400 03 0.21824380 0	ľg
100	271.5	715.16	2.3549	43.45	0-118031070	20	A.20446070 C3 U.2U056UIUU V3 V.2IAT04470 A	2
	271-0				-0+114246910	3	0 2033670 00-01 10 100000000 00 00 000000000	5
	270.5	704.50	£502°3	43+95	0-111544340	3	0.202274520 U3 0.204357010 U 0.204142340 01	2
	270-0			- 44.01	C5011E801+0	3	0.20113354U -0	
-	269.5	69.623	2.0718	44.33	0.1053E533D	ĉ	0.200108465 03 0.2021/203L V3 V.2VV3207L V3 	
000	- 0-095	669+75		50.44	-0-112460942JL-0-	*	C UTAT IAAAC A TA ULAUNUUUUUUUUUUUUUUUUUUUUUUUUUUUUUUUUU	
5907	24845	664.71	1.9417	44.91	0*955364020	10	0.19795964D 03 0.200046240 03 0.207497 0	
	2.664	675.63	1.8774	45.17	0.9661216.0	5	C.15666680 03 C.198424090 US V.2V37477 0	38
	267.45	624-67	1.6135	45.40	0.936882910	5	0+195814310 03 0+197542750 03 0+201470 0	5
	2664			90°*94-	0-532448270	た	0.194742720 U. 0.19664390 0 0.20074242400 0	
005 - 1	266.4	664.69	1.6673	45.75	3.87841801 U	1		
	265.0			59-54	0*84070544D	7	U UVJ961001 V EV UVDVILVO 10 005109261*0-	20
117	265.4	654.75	1.5623	45.36	015125619*0	5	0.19153280D 03 0.15325350 03 0.15305350 0	8
	264.8			- A6.CA	348132152*0	7	0.190465650 0. 0. 0.19046900 0. 0.10001300 0	6
953.5	264.3	644.67	1.4687	47.09	16 19 16 144.0	10	3.183428160 03 0.191401/10 03 0.1940204100 03 0.104020400.0	3
	262.2				-0-76326365	đ	D USULANES . V EV GESULANE V EV GESULANE V EV GESULANES . V EV GESULANE V	8
3-65 0	262.1	635+05	1.40.33	45.25	0.749212340	2	0 01092020100 FM JEECSCORE 0 20 20000000000000000000000000000000	3
212	265.5	630+16	1-3709	50.37	-0+735144230	5	0.150318820-03-04-1882561-04-04-04-04-04-04-04-04-04-04-04-04-04-	C
	262.0	625.25	1.2387	51.51	0.721065530	5	0.165283440 03 0.167217480 03 VII31110010 	3
		-14009			0-766276230	3	0-184 245010	0
464° E	200.5	615.56	1.2750	53 - 86	0.692876430	5 0	0-183214350 03 V.153126950 03 0-18643240-0	3
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U.22] 330245 V2 V.22375355 V2 V22375355 V2 V223420450 V2	0.0 0.3	ú•0	04.00	211-5	19.130
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24.505	214.1	1 6.76	5. 5			
24-652						9.642795060 01 0.649768370 01 0.146561360 01
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		12.24			q	0.606903370 01 0.613427030 01 0.610140480 01
52.73	244 2	16.75	0.0			0.589493960 01 0.59444444 01 0.592504540 01
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27.591	214.3	15.60	0 0 0			U.555755190 UL VOOLETEN 01 0.558295220 01
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365.05	214.5		0.0	0.0	0.0	0.330 EVELT 01 0.34783871F 01 0.356195050 UI
20-577	215.6			0.0	0.0	0.344139368 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
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		346		0-0	0.0	C.257679780 01 V.20077501 01. 0.256820780-01
32.550	216.7	7.19		0.0	0.0	
-554-52	-9-916	-25-9			0.0	0.23896207D 01 U. CTTTESUE 1 0.237003AD 01
23-072	217.1	6.69	0.0	2.0		0.223848410 01 0.434444 01 0.228829490 01
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33.555	237.4	6•1 5				0.212635959 01 0.4 1 00 01 0.211640220 01
33-807	-0. Cta	Se 56			0.0	0-204477190 01 0-2000 01 0-203453410 0
34.058	217.6	5.73	0.0		0-0	0+146577130 01 0+14654955 01 0-155509480 0
ELE-AD	-9-746				0-0	0.18869240D 01 0.19972291 01 0.18777494D 0
34.568	217.6	2• %				0. 161 419630 -01 -01 -01 -01 -01 01 0-180256330 0
34.828				0.0	0.0	0.174155490 UI VALVATOR 0.173950190 C
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35+358-				0.0	0.0	6-16023974D 01 0-1553316D 01 0-15896160D 0
35.628	217.6			4	-0-0	
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0°J \$0°0 0°J £90 0°J £90	0.01 0.01 0.02 0.0 0.01 0.0 0.01 0.0	0.01 0.00 0.01 0.0 0.01 0.0 0.00 0.0	0*0 C*0	0.0 0.1 0.1 0.0 0.0 0.0 0.0 0.0	0*0 0*0 0*0 0*3 0*0 0*0
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STATION # 92734 YEAR # 67 MONTH # 1 LAY # 1 HOUP # 12 -------ELEVATION ANGLE RANGE = 0.277030160 04 KILOMETERS SUM OF HEFRACTIVITY - Asp3660JOD-04 HEIGHT = 0.845872630-01 KILCHETERS PHESEWRE -----TEMPERATURE = -3.42000000 01 DEGREES CELCIUS RELATIVE HUMIDITY = - 0.55000000 02-04 CENT -ELEVATION ANGLE ENROR = 0.157860250-02 RADIANS STATION = 93734 YEAR = 67 MCNTH = 1 DAY = 1 HOUR = 12 STATION - STON TEAR - OF MUNITI - I DAT - I HOW - IE ELEVATION ANGLE - 0.241355400. GO RADIANS - ---RANGE = 0.241261120 04 KILOMETERS SUM OF FEFRACTIVITY 0.236660300-04----HEIGHT = 0.845872630-01 KILOMETERS TENPERATURE = -0.420000000 01 DEGREES CELCIUS RELATING MUNICITY -- 0. SECONDED 02 PERCENT ELEVATION ANGLE ERROR = 0.106085630-02 HADIANS RANGE - ERROR APPEUX- - - 0.502288920-01 #F7585 RANGE DIFF = 0.121481560 00 CM STATION = 93734 YEAR = 67 MCNTH = 1 EAY = 1 HOUR = 12 ELEVATION ANGLE = 0.349068860 CO RADIANS RANGE # 0.212377556 04 KILCHETERS مراجعة ومصفح المراجع المراجع الراري والأراج MEIGHT = 0.349672630-01 KILCMETERS PRESSURE - 0.100300000 U4-MILLIPARS TEMPERATURE = -9.42000000 01 DEGREES CELCIUS RELATING HUMEDITY - 0.550COCOD 02 .PERCENT FLEVATION ANGLE ERROR = 0.787449380-03 RADIANS RANGE ERROR APPERATE -0+687243010 -01 -MLTERS RANGE DIFF = -C.623289140-01 CM STATION = \$3734 YEAH = 67 WENTH = 1 CAY = 1 HOUR = 12 RANGE = 0.142916550 C4 KILCHE TERS HEIGHT = C.84987263D-01 KILOMETERS TEMPERATURE = -0.420000000 01 LEGREES CELCIUS ELEVATION ANGLE ERROR = 0.344544930-03 PADIANS RANGE ERPER APPROX - 0. 267403700 01 METERS RANGE CIFF = -C. 140334050 CO CM STATION = 93734 YEAR = 07 NONTH = 1 CAY = 1 HOUR = 12 ELEVATION ANGLE - ON JOSSAND 31-RADIANS RANGE = 0.101320360 04 KILOMETERS HEIGHT = 0.845872830-01 KILOMETERS PRESSURE - 0-100 JOGGOD OA MILLIBARS TEMPERATURE = -0.420000000 01 DEGREES CELCIUS ELEVATION ANGLE EHECR # 0.510602370-CA HALIANS RANGE ERER APRENT ... 0.240227440 .01 METURS RANGE CIFF = -0.107730840 CO CM

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