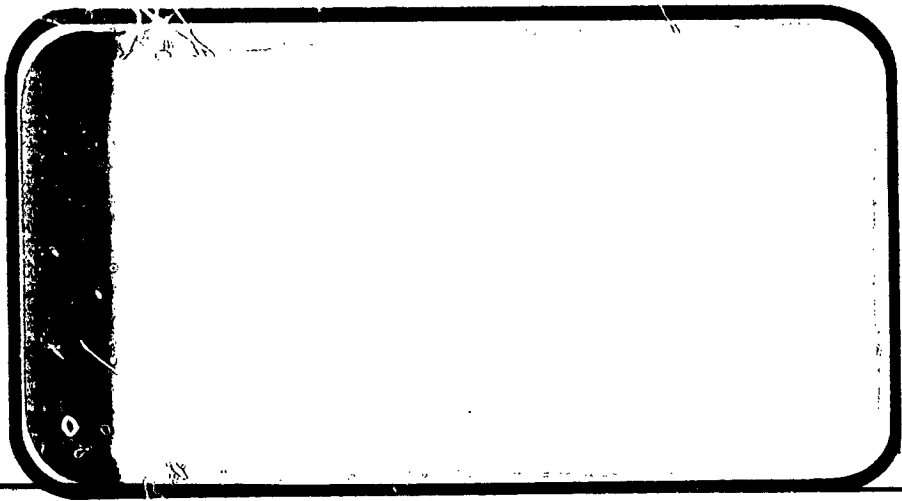




# NATIONAL AERONAUTICS AND SPACE ADMINISTRATION



(NASA-CR-128795) WIND TUNNEL TEST OF  
THE 0.010-SCALE SPACE SHUTTLE INTEGRATED  
VEHICLE IN THE NASA-AMES 3.5 FOOT  
HYPERSONIC WIND TUNNEL (IA10) (Chrysler  
Corp.) 138 p HC \$9.00 CSCI 22B

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SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT

JOHNSON SPACE CENTER

HOUSTON, TEXAS

DATA Management services

SPACE DIVISION



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WIND TUNNEL TEST OF THE 0.010-SCALE  
SPACE SHUTTLE INTEGRATED VEHICLE IN THE  
NASA-AMES 3.5-FOOT HYPERSONIC WIND TUNNEL (IA10)

By

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National Aeronautics and Space Administration  
Houston, Texas

WIND TUNNEL TEST SPECIFICS:

Test Number: ARC 3.5-169  
NASA Series No.: IA10  
Test Dates: 1 thru 3 August 1973

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WIND TUNNEL TEST OF THE 0.010-SCALE  
SPACE SHUTTLE INTEGRATED VEHICLE IN THE  
NASA-AMES 3.5-FOOT HYPERSONIC WIND TUNNEL (IA10)

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J. W. Cleary and J. A. Mellenthin/NASA Ames

ABSTRACT

Experimental aerodynamic investigations were conducted in the NASA Ames Research Center 3.5-Foot Hypersonic Wind Tunnel from August 1, 1973 to August 3, 1973 on a 0.010-scale model of the Space Shuttle Vehicle orbiter and external tank (model no. 32 OT).

The purpose of the test was to evaluate the basic hypersonic stability characteristics of the external tank and orbiter and to define orbiter plume effects on aero characteristics using solid plumes.

The test was conducted at angles of attack from  $-10^{\circ}$  to  $30^{\circ}$  and angles of sideslip of  $-10^{\circ}$  thru  $10^{\circ}$ . Six component force data and static base pressures were recorded during the test.

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COEFFICIENTS SCHEDULE:

A: CA, CN, CIM, vs. ALPHA

CN vs. CIM

B: CY, CBL, CYN vs. BETA

CY vs. CYN

C: CY, CBL, CYN vs. ALPHA

D: CN/A, CIM/A, XAC/L vs. ALPHA

DCY/DA, DCELDA, DCYNDA, DCN/DA, DCIMDA, DCA/DA vs. ALPHA

E: DCY/DB, DCELDB, DCYNDB vs. ALPHA

NOMENCLATURE  
General

<u>SYMBOL</u>	<u>PLOT SYMBOL</u>	<u>DEFINITION</u>
a		speed of sound; m/sec, ft/sec
$C_p$	CP	pressure coefficient; $(p_1 - p_\infty)/q$
M	MACH	Mach number; $V/a$
p		pressure; $N/m^2$ , psf
q	Q(NSM)	dynamic pressure; $1/2\rho V^2$ , $N/m^2$ , psf
RN/L	RN/L	unit Reynolds number; per m, per ft
V		velocity; m/sec, ft/sec
$\alpha$	ALPHA	angle of attack, degrees
$\beta$	BETA	angle of sideslip, degrees
$\psi$	PSI	angle of yaw, degrees
$\phi$	PHI	angle of roll, degrees
$\rho$		mass density; $kg/m^3$ , slugs/ft <sup>3</sup>

Reference & C.G. Definitions

c.g.		center of gravity
$l_{REF}$	LREF	reference length or wing mean aerodynamic chord; m, ft
S	SREF	wing area or reference area; m <sup>2</sup> , ft <sup>2</sup>
	MRP	moment reference point
	XMRP	moment reference point on X axis
	YMRP	moment reference point on Y axis
	ZMRP	moment reference point on Z axis

NOMENCLATURE (Continued)

SUBSCRIPTS

b	base
l	local
s	static conditions
t	total conditions
	free stream

Body-Axis System

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
$C_N$	CN	normal-force coefficient; normal force/qS
$C_A$	CA	axial-force coefficient; axial force/qS
$C_Y$	CY	side-force coefficient; side force/qS
$C_m$	CIM	pitching-moment coefficient; pitching moment/qS $l$
$C_n$	CYN	yawing-moment coefficient; yawing moment/qS $l$
$C_l$	CBL	rolling-moment coefficient; rolling moment/qS $l$

Stability-Axis System

$C_L$	CL	lift coefficient; lift/qS
$C_D$	CD	drag coefficient; drag/qS
L/D	L/D	lift-to-drag ratio; $C_L/C_D$

ADDITIONS TO NOMENCLATURE

$C_{N\alpha_{local}}$	CN/A	local normal force coefficient derivative with alpha; per degree.
$C_{m\alpha_{local}}$	CIM/A	local pitching moment coefficient derivative with alpha; per degree.
	XAC/L	local longitudinal center of pressure.

NOMENCLATURE (Continued)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
$C_{Y\delta a}$	DCY/DA	side force coefficient derivative with respect to total aileron deflection. Algebraic difference of the side force coefficients of two runs divided by the algebraic difference of the total aileron deflection angle of the runs; per degree.
$C_{l\delta a}$	DCBLDA	rolling moment coefficient derivative with respect to total aileron deflection. Algebraic difference of the rolling moment coefficient of two runs divided by the algebraic difference of the total aileron deflection angle of the runs; body axis system; per degree.
$C_{n\delta a}$	DCYNDA	yawing moment coefficient derivative with respect to total aileron deflection. Algebraic difference of the yawing moment coefficient of two runs divided by the algebraic difference of the total aileron deflection angle of the runs; body axis system; per degree.
$C_{N\delta a}$	DCN/DA	normal force coefficient derivative with respect to aileron deflection. Algebraic difference of normal coefficient of two runs divided by the algebraic difference of the total aileron deflection of the runs; body axis system; per degree.
$C_{m\delta a}$	DCIMDA	pitching moment coefficient derivative with respect to aileron deflection. Algebraic difference of pitching moment coefficient of two runs divided by the algebraic difference of the total aileron deflection of the runs; body axis system; per degree.
$C_{A\delta a}$	DCA/DA	axial force coefficient derivative with respect to aileron deflection. Algebraic difference of axial force of two runs divided by the algebraic difference of total aileron deflection of the runs; body axis system; per degree.
$C_{Y\beta}$	DCY/DB	side force coefficient derivative with respect to beta. Algebraic difference of the side force coefficient of two runs divided by the algebraic difference of the side slip angle of the runs; per degree.

NOMENCLATURE (Continued)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
$C_{l\beta}$	DCBLDB	rolling moment coefficient derivative with respect to beta. Algebraic difference of the rolling moment coefficient of two runs divided by the algebraic difference of the side slip angle of the runs; body axis system; per degree.
$C_{n\delta r}$	DCYNDB	yawing moment coefficient derivative with respect to beta. Algebraic difference of the yawing moment coefficient of two runs divided by the algebraic difference of the side slip angle of the runs; per degree.
$C_{N\delta r}$	DCN/DR	normal force coefficient derivative with respect to rudder deflection. Algebraic difference of normal force coefficient of two runs divided by algebraic difference of rudder deflection of two runs; per degree.
$C_{Y\delta r}$	DCY/DR	side force coefficient derivative with respect to rudder deflection. Algebraic difference of the side force coefficient of two runs divided by the algebraic difference of the rudder deflection angle of the runs; body axis system; per degree.
$C_{n\delta r}$	DCYNDR	yawing moment coefficient derivative with respect to rudder deflection. Algebraic difference of the yawing moment coefficient of two runs divided by the algebraic difference of the rudder deflection angle of the runs; body axis system; per degree.
$C_{l\delta r}$	DCBLDR	rolling moment coefficient derivative with respect to rudder deflection. Algebraic difference of the rolling moment coefficient of two runs divided by the algebraic difference of the rudder deflection angle of the runs; body axis system; per degree.
$C_{A\delta r}$	DCA/DR	axial force coefficient derivative with respect to rudder deflection. Algebraic difference of axial force coefficient of two runs divided by the algebraic difference of rudder deflection of two runs; per degree.

NOMENCLATURE (Continued)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
$C_{m_{\delta_r}}$	DCLMDR	pitching moment coefficient derivative with respect to rudder deflection. Algebraic difference of pitching moment coefficient of two runs divided by algebraic difference of rudder deflection of two runs; per degree.
	YAC/L	lateral center of pressure. Yawing moment coefficient derivative divided by side force coefficient derivative.
$C_{A_{\alpha=0}}$	CAAFO	axial force coefficient at zero angle of attack (ALPHA = 0).
$C_{N_{\alpha=0}}$	CNAFO	normal force coefficient at zero angle of attack (ALPHA = 0).
$C_{m_{\alpha=0}}$	CLMAFO	pitching moment coefficient at zero angle of attack (ALPHA = 0).
$\delta_{eL}$		left elevon surface deflection angle, positive deflection trailing edge down; degrees.
$\delta_{eR}$		right elevon surface deflection angle, positive deflection trailing edge down; degrees.
$\delta_e$	ELEVON	elevon, surface deflection angle, positive deflection trailing edge down; degrees, $(\delta_{eL} + \delta_{eR})/2$ .
$\delta_a$	AILRON	aileron, surface deflection angle, positive deflection trailing edge down; degrees $(\delta_{eL} + \delta_{eR})/2$ .
$\delta_r$	RUDDER	rudder, surface deflection angle, positive deflection trailing edge to the left; degrees.
$A_{BC}$		external tank balance cavity area, 3.999 in <sup>2</sup> .
$A_{ET}$		external tank base area, 4.255 in <sup>2</sup> model scale.

## NOMENCLATURE (Concluded)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
A <sub>ME</sub>		orbiter main engine nozzle base area, 2.081 in <sup>2</sup> model scale.
A <sub>OB</sub>		orbiter base area, 2.010 in <sup>2</sup> model scale.
A <sub>OM</sub>		OMS pod base area, 1.362 in <sup>2</sup> model scale.
C <sub>A<sub>b</sub></sub>	CAB	orbiter base axial-force coefficient.
C <sub>A<sub>b</sub>ET</sub>	CABET	external tank base axial-force coefficient.
C <sub>PBC</sub>	CPBC	external tank balance cavity pressure coefficient.
C <sub>PET</sub>	CPET	external tank base pressure coefficient.
C <sub>POB</sub>	CPOB	orbiter base pressure coefficient.
C <sub>POM</sub>	CPOM	OMS pod base pressure coefficient.
C <sub>PME</sub>	CPME	main engine nozzle base pressure coefficient.

## CONFIGURATIONS INVESTIGATED

The model for this test was an 0.010-scale representation of the Space Shuttle Vehicle configuration 3 Space Shuttle (model no. 32 OT). The model consisted of the orbiter, external tank and simulated engine plumes. The model was constructed of stainless steel except the orbiter body which was constructed of aluminum.

A six component 1.5-inch MK II balance was mounted in the external tank.

The model was tested with and without non-metric main engine solid exhaust plume simulators. Figure 2(c) shows the model with plume simulation installed.

The various model components tested are listed below:

$O_9 = B_{19} C_7 E_{23} F_5 M_4 N_{24} N_8 R_5 V_7 W_{107}$

$B_{19} =$  VL70 - 000139B (lines) body

$C_7 =$  VL70 - 000139B (lines) canopy

$E_{23} =$  VL70 - 000139B (lines) elevon

$F_5 =$  VL70 - 000139B (lines) body flap

$M_4 =$  VL70 - 000139B (lines) OMS pod

$N_{24} =$  VL70 - 000140A (lines) main engine nozzle

$N_8 =$  VL70 - 000140A (lines) OMS nozzle

$R_5 =$  VL70 - 000139B and VL70 - 000095 rudder

$V_7 =$  VL70 - 000139B (lines) vertical tail



W<sub>107</sub> = VL70 - 000139B (lines) wing  
T<sub>10</sub> = VL78 - 000041 (lines) external tank  
AT<sub>2</sub> = VL72 - 000089 (lines) attach structure

Base pressure data acquired during runs 1 thru 6 are invalid. Tygon tubing, used in measuring these pressures was melted by the tunnel heat during these runs. Prior to run five two external pressure tubes were added. The location of these pressure tubes is shown in figure 2(b).

Pressure tap #16 has an area value of 4.378 in<sup>2</sup> and pressure tap #17 has an area value of 2.010 in<sup>2</sup>. Subsequent to run 006 tygon tubing was replaced, in the orbiter, with a high temperature resistant flexible tubing. With the exception of tap #4 which was inoperative, the data acquired from that time to the end of the run series are considered reliable.

## TEST FACILITY DESCRIPTION

The NASA-Ames 3.5-Foot Hypersonic Wind Tunnel is a closed-circuit, blowdown-type tunnel capable of operating at nominal Mach numbers of 5, 7, and 10 at pressures to 1800 psia and temperatures to 3400°R for run times to four minutes. The major components of the facility include a gas storage system where the test gas is stored at 3000 psi, a storage heater filled with aluminum-oxide pebbles capable of heating the test gas to 3400°R, axisymmetric contoured nozzles with exit diameters of 42 inches for generating the desired Mach number, and a 900,000 ft<sup>3</sup> vacuum storage system which operates to pressures of 0.3 psia. The test section itself is an open-jet type enclosed within a chamber approximately 12-feet in diameter and 40-feet in length, arranged transversally to the flow direction.

A model support system is provided that can pitch models through an angle-of-attack range of -20 to +18 degrees, in a vertical plane, about a fixed point of rotation on the tunnel centerline. This rotation point is adjustable from 1 to 5 feet from the nozzle exit plane. The model normally is out of the test stream (strut centerline 37-inches from tunnel centerline) until the tunnel test conditions are established after which it is inserted. Insertion time is adjustable to as little as 1/2 second and models may be inserted at any strut angle.

A high-speed, analog-to-digital data acquisition system is used to record test data on magnetic tape. The present system is equipped to measure and record the outputs from 80 transducers in addition to 20 channels of tunnel parameters.

## DATA REDUCTION

The aerodynamic force data presented were measured by the Task Corporation 1.5- inch MK II strain gage balance and the moment data were transferred to the external tank centerline at a point 6.80 in (model scale) from the tank nose.

Base pressure axial-force coefficients were calculated for the individual regions as follows:

1. Orbiter Base Axial Force

$$C_{A_b} = - (A_{ME} C_{P_{ME}} + A_{OB} C_{P_{OB}} + A_{OM} C_{P_{OM}}) / S$$

2. External Tank Base Axial Force

$$C_{A_b_{ET}} = - (A_{ET} C_{P_{ET}} + A_{BC} C_{P_{BC}}) / S$$

where S is the theoretical wing area (0.269 ft<sup>2</sup> model scale).

These base pressure axial-force coefficients were calculated and plotted but were not used to correct the measured axial-force coefficient data.

The following reference dimensions were used for data reduction:

Symbol	Description	Value (model scale)	Applicable Data
$A_{ME}$	Orbiter main engine nozzle base area	2.081 in <sup>2</sup>	Pressure tap nos. 3 and 4
$A_{OB}$	Orbiter base area	2.010 in <sup>2</sup>	Pressure tap nos. 5 and 6 or 17
$A_{OM}$	OMS pod base area	1.362 in <sup>2</sup>	Pressure tap nos. 1 and 2

DATA REDUCTION (Concluded)

<u>Symbol</u>	<u>Description</u>	<u>Value (model scale)</u>	<u>Applicable Data</u>
$A_{ET}$	ET base area	4.255 in <sup>2</sup>	Pressure tap nos. 7 and 8 or 16
$A_{BC}$	ET balance cavity area	3.990 in <sup>2</sup>	Pressure tap no. 15
S	Reference area	0.269 ft <sup>2</sup>	Force and moment data
$l_r$	Reference length	12.90 in	Moment data

Also see discussion in section "CONFIGURATIONS INVESTIGATED" concerning the use of external pressure taps 16 and 17.





TABLE 3. - MODEL COMPONENT DESCRIPTIONS

MODEL COMPONENT: BODY - B19

GENERAL DESCRIPTION: Fuselage, Configuration 3, per Rockwell Lines  
VL70-000139B.

NOTE: Identical to B17 except forebody.

Model Scale = .010

DRAWING NUMBER: VL70-000139B

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length - IN.	<u>1290.3</u>	<u>12.903</u>
Max. Width - IN.	<u>267.6</u>	<u>2.676</u>
Max. Depth - IN.	<u>244.5</u>	<u>2.445</u>
Fineness Ratio	<u>4.82175</u>	<u>4.82175</u>
Area - FT <sup>2</sup>		
Max. Cross-Sectional	<u>386.67</u>	<u>0.03867</u>
Planform	<u>          </u>	<u>          </u>
Wetted	<u>          </u>	<u>          </u>
Base	<u>          </u>	<u>          </u>

TABLE 3. - Continued.

MODEL COMPONENT: Canopy - C7

GENERAL DESCRIPTION: Configuration 3 per Rockwell Lines VL70-000139

Model Scale = .010

DRAWING NUMBER VL70-000139

<u>DIMENSION:</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Length ( $X_0 = 433$ to $X_0 = 670$ ) - in. FS	<u>237</u>	<u>2.37</u>
Max Width	<u>          </u>	<u>          </u>
Max Depth ( $Z_0 =$ to $Z_0 = 501$ ) - in FS	<u>          </u>	<u>          </u>
Fineness Ratio	<u>          </u>	<u>          </u>
Area		
Max Cross-Sectional	<u>          </u>	<u>          </u>
Planform	<u>          </u>	<u>          </u>
Wetted	<u>          </u>	<u>          </u>
Base	<u>          </u>	<u>          </u>



TABLE 3. - Continued.

MODEL COMPONENT: ELEVON - E23

GENERAL DESCRIPTION: Configuration 3 per W107 Rockwell Lines

VL70-000139B, data for (1) of (2) sides

Model Scale = .010

DRAWING NUMBER: VL70-000139B

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area - FT <sup>2</sup>	<u>205.52</u>	<u>0.02055</u>
Span (equivalent) - IN.	<u>353.34</u>	<u>3.5334</u>
Inb'd equivalent chord	<u>114.78</u>	<u>1.1478</u>
Outb'd equivalent chord	<u>55.00</u>	<u>0.550</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>.208</u>	<u>.208</u>
At Outb'd equiv. chord	<u>.400</u>	<u>.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>0.00</u>	<u>0.00</u>
Tailing Edge	<u>-10.24</u>	<u>-10.24</u>
Hingeline	<u>0.00</u>	<u>0.00</u>
Area Moment (Normal to hinge line) - FT <sup>3</sup>	<u>1548.07</u>	<u>0.00155</u>
Product of Area Moment		

TABLE 3. - Continued.

MODEL COMPONENT: F5 Body Flap

GENERAL DESCRIPTION: 3 Configuration per Rockwell Lines VL70-000139

Scale Model = .010

DRAWING NUMBER

VL70-000139

DIMENSION:

FULL SCALE

MODEL SCALE

Length - in

84.70

0.8470

Max Width - in

267.6

2.676

Max Depth

Fineness Ratio

Area - Ft<sup>2</sup>

Max Cross-Sectional

Planform

142.00

0.0142

Wetted

Base

38.0958

0.00381

TABLE 3. - Continued.

MODEL COMPONENT: O/S Pod - M<sub>4</sub>

GENERAL DESCRIPTION: Configuration 3 per Rockwell Lines VL70-000139

NOTE: M<sub>4</sub> identical to M<sub>3</sub>, except intersection to fuselage.

Model Scale = .010.

DRAWING NUMBER

VL70-000139

DIMENSION:

FULL SCALE

MODEL SCALE

Length - IN

346.0

3.460

Max Width - IN

108.0

1.080

Max Depth - IN

113.0

1.130

Fineness Ratio

Area - FT<sup>2</sup>

Max Cross-Sectional

Planform

Wetted

Base

TABLE 3. - Continued.

MODEL COMPONENT: Attach Structure - AT<sub>2</sub>

GENERAL DESCRIPTION: Attach Structure for Orbiter-Tank Configuration 3 per

Rockwell Lines VL70-000139B, VL78-000041

MODEL SCALE = 0.010

DRAWING NO. SS-A00060

DIMENSIONS:

FORWARD ATTACH POINTS

Orbiter to Tank

Number of Struts

Diameter in.

Location in.

X<sub>O</sub>

X<sub>T</sub>

FULL SCALE

MODEL SCALE

1

1

25.000

0.250

382.000

3.820

1077.000

10.770

Orbiter to SRB

Number of Struts

Diameter in.

Location in.

X<sub>O</sub>

X<sub>S</sub>

Tank to SRB

Number of Struts

Diameter in.

Location in.

X<sub>T</sub>

X<sub>S</sub>

AFT ATTACH POINTS

Orbiter to Tank

Number of Struts /Side

Diameter in.

Location in.

X<sub>O</sub>

X<sub>T</sub>

3

3

15.000

0.150

1292, 1308, 1308

12.92, 13.08, 13.08

1859, 2061, 2061

18.59, 20.61, 20.61

Orbiter to SRB

Number of Struts

Diameter in.

Location in.

X<sub>O</sub>

X<sub>S</sub>

Tank to SRB

Number of Struts

Diameter in.

Location in.

X<sub>T</sub>

X<sub>S</sub>

TABLE 3. - Continued.

MODEL COMPONENT: NOZZLES - N g

GENERAL DESCRIPTION: Basic OMS nozzle of Configuration 2A per Rockwell Lines

VL70-008306 and VL70-000089"B". Intersection of nozzle exit plane and  
nozzle centerline at  $X_0 = 1570.75$ ,  $Y_0 = +99.25$ ,  $Z_0 = 507.25$

MODEL SCALE = .010

DRAWING NO. VL70-008306, VL70-000089"B", SS-A00092

<u>DIMENSIONS</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Mach No. _____		
Length ~ in.		
Gimbal Point to Exit Plane	_____	_____
Throat to Exit Plane	_____	_____
Diameter ~ in.		
Exit	<u>50.00</u>	<u>0.500</u>
Throat	<u>N/A</u>	<u>N/A</u>
Inlet	<u>28.00</u>	<u>0.280</u>
Area ~ ft <sup>2</sup> /Nozzle		
Exit	<u>13.635</u>	<u>0.00136</u>
Throat	_____	_____
Gimbal Point (station) ~ in.		
X	<u>1518.0</u>	<u>15.18</u>
Y	<u>+88.0</u>	<u>+0.88</u>
Z	<u>492.0</u>	<u>4.92</u>
Null Position ~ deg.		
Pitch	<u>15°49'</u>	<u>15°49'</u>
Yaw (Outb'd)	<u>±12°17'</u>	<u>±12°17'</u>

TABLE 3. - Continued.

MODEL COMPONENT: MPS NOZZLES - N 24

GENERAL DESCRIPTION: Configuration 3A MPS Nozzles

MODEL SCALE = .010

DRAWING NO. VL70-000140A, VL70-005030A

<u>DIMENSIONS</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Mach No. _____		
Length ~ in.		
Gimbal Point to Exit Plane	_____	_____
Throat to Exit Plane	_____	_____
Diameter ~ in.		
Exit	<u>91.000</u>	<u>0.910</u>
Throat	_____	_____
Inlet	_____	_____
Area ~ ft <sup>2</sup> . / Nozzle		
Exit	<u>45.16585</u>	<u>0.00453</u>
Throat	_____	_____
Gimbal Point (station) ~ in.		
Upper Nozzle		
X	<u>14.45</u>	<u>14.45</u>
Y	<u>0</u>	<u>0</u>
Z	<u>4.43</u>	<u>4.43</u>
Lower Nozzles		
X	<u>1468.16996</u>	<u>14.68170</u>
Y	<u>153.00000</u>	<u>0.53000</u>
Z	<u>342.63988</u>	<u>3.42640</u>
Null Position ~ deg.		
Upper Nozzle		
Pitch	<u>16°</u>	<u>16°</u>
Yaw	<u>0°</u>	<u>0°</u>
Lower Nozzles		
Pitch	<u>10°</u>	<u>10°</u>
Yaw (outb'd)	<u>3.5°</u>	<u>3.5°</u>

TABLE 3. - Continued.

MODEL COMPONENT: Solid Plume - PL 1

GENERAL DESCRIPTION: SSME simulated plumes from N24 nozzles to  
represent all 3 engines at M= 5.5 during exit trajectory

MODEL SCALE = .010

DRAWING NUMBER: \_\_\_\_\_

COORDINATES:

Ratio of local plume radius  
to nozzle exit plane internal  
radius

Ratio of local axial distance  
from nozzle exit plane to nozzle  
exit plane internal radius

<u>1.053</u>	<u>0.057</u>
<u>1.943</u>	<u>1.122</u>
<u>2.772</u>	<u>2.250</u>
<u>3.497</u>	<u>3.341</u>
<u>4.450</u>	<u>4.912</u>
<u>5.421</u>	<u>6.642</u>
<u>5.905</u>	<u>7.566</u>
<u>6.389</u>	<u>8.529</u>
<u>7.321</u>	<u>10.496</u>
<u>7.861</u>	<u>11.699</u>
<u>8.136</u>	<u>12.330</u>
<u>8.672</u>	<u>13.602</u>
<u>8.937</u>	<u>14.367</u>
<u>9.204</u>	<u>14.912</u>
<u>9.464</u>	<u>15.569</u>

DIMENSIONS:  
Nozzle Exit Radius, in.

FULL SCALE  
45.2

MODEL SCALE  
0.452

TABLE 3. - Continued.

MODEL COMPONENT: EXTERNAL TANK - T10

GENERAL DESCRIPTION: External Oxygen Hydrogen Tank, 3 Configuration,  
per Rockwell Lines VL78-000041 and VL72-000088

Model Scale = .010

DRAWING NUMBER VL72-000088  
VL78-000041

<u>DIMENSION:</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Length - IN (Nose @ $X_T = 309$ )	<u>1865</u>	<u>18.65</u>
Max Width (Dia) - IN.	<u>324</u>	<u>3.24</u>
Max Depth	<u>-</u>	<u>-</u>
Fineness Ratio	<u>5.75617</u>	<u>5.75617</u>
Area - FT <sup>2</sup>		
Max Cross-Sectional	<u>572.555</u>	<u>0.05726</u>
Planform	<u>          </u>	<u>          </u>
Wetted	<u>          </u>	<u>          </u>
Base	<u>          </u>	<u>          </u>
WP of Tank Centerline ( $X_T$ ) IN.	<u>400.0</u>	<u>4.00</u>



TABLE S. - Continued.

MODEL COMPONENT: RUDDER - R5

GENERAL DESCRIPTION: 2A, 3 and 3A Configuration per Rockwell Lines

VL70-000095

Model Scale = .010

DRAWING NUMBER: VL70-000095

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area - FT <sup>2</sup>	<u>106.38</u>	<u>0.01064</u>
Span (equivalent) - IN.	<u>201.0</u>	<u>2.010</u>
Inb'd equivalent chord	<u>91.585</u>	<u>0.91585</u>
Outb'd equivalent chord	<u>50.833</u>	<u>0.50833</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
At Outb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>34.83</u>	<u>34.83</u>
Tailing Edge	<u>26.25</u>	<u>26.25</u>
Hingeline	<u>34.83</u>	<u>34.83</u>
Area Moment (Normal to hinge line)- FT <sup>3</sup>	<u>526.13</u>	<u>0.00053</u>
Product of Area and Mean Chord		

TABLE 3. - Continued.

MODEL COMPONENT: VERTICAL - V7

GENERAL DESCRIPTION: Centerline vertical tail, double-edge airfoil with rounded leading edge.

NOTE: Same as V5, but with manipulator housing removed.

Model Scale = .010

DRAWING NUMBER:

VL70-000139

DIMENSIONS:

FULL SCALE

MODEL SCALE

TOTAL DATA

Area (Theo) Ft <sup>2</sup>	<u>425.92</u>	<u>0.04259</u>
Planform		
Span (Theo) In	<u>315.72</u>	<u>3.157</u>
Aspect Ratio	<u>1.675</u>	<u>1.675</u>
Rate of Taper	<u>0.507</u>	<u>0.507</u>
Taper Ratio	<u>0.404</u>	<u>0.404</u>
Sweep Back Angles, degrees		
Leading Edge	<u>45.000</u>	<u>45.000</u>
Trailing Edge	<u>26.249</u>	<u>26.249</u>
0.25 Element Line	<u>41.130</u>	<u>41.130</u>
Chords:		
Root (Theo) WP	<u>268.50</u>	<u>2.6850</u>
Tip (Theo) WP	<u>108.47</u>	<u>1.0847</u>
MAC	<u>199.81</u>	<u>1.9981</u>
Fus. Sta. of .25 MAC	<u>1463.50</u>	<u>14.6350</u>
W. P. of .25 MAC	<u>635.522</u>	<u>6.3552</u>
B. L. of .25 MAC	<u>0.00</u>	<u>0.00</u>
Airfoil Section		
Leading Wedge Angle Deg	<u>10.000</u>	<u>10.000</u>
Trailing Wedge Angle Deg	<u>14.920</u>	<u>14.920</u>
Leading Edge Radius	<u>2.0</u>	<u>0.020</u>
Void Area - Ft <sup>2</sup>	<u>13.17</u>	<u>0.00132</u>
Blanketed Area	<u>0.00</u>	<u>0.00</u>

TABLE 3. - Concluded.

MODEL COMPONENT: WING-W 107

GENERAL DESCRIPTION: Configuration 3 per Rockwell Lines VI70-000139B

NOTE: Same as W103, except cuff, airfoil and incidence angle.

Model Scale = .010

TEST NO.

DWG. NO. VI70-000139B

DIMENSIONS:

FULL-SCALE

MODEL SCALE

TOTAL DATA

Area (Theo.) Ft<sup>2</sup>

Planform

2690.00

0.2690

Span (Theo In.

936.68

9.3668

Aspect Ratio

2.265

2.265

Rate of Taper

1.177

1.177

Taper Ratio

0.200

0.200

Dihedral Angle, degrees (@ TE of Elevon)

3.500

3.500

Incidence Angle, degrees

0.500

0.500

Aerodynamic Twist, degrees

+3.000

+3.000

Sweep Back Angles, degrees

Leading Edge

45.000

45.000

Trailing Edge

-10.24

-10.24

0.25 Element Line

35.209

35.209

Chords:

Root (Theo) B.P.O.O.

689.24

6.8924

Tip, (Theo) B.P.

137.85

1.3785

MAC

474.81

4.7481

Fus. Sta. of .25 MAC

1136.89

11.3689

W.P. of .25 MAC

299.20

2.9920

B.L. of .25 MAC

182.13

1.8213

EXPOSED DATA

Area (Theo) Ft<sup>2</sup>

1752.29

0.17523

Span, (Theo) In. BP108

720.68

7.2068

Aspect Ratio

2.058

2.058

Taper Ratio

0.2451

0.2451

Chords

Root BP108

562.40

5.6240

Tip 1.00  $\frac{b}{2}$

137.85

1.3785

MAC

393.03

3.9303

Fus. Sta. of .25 MAC

1185.31

11.8531

W.P. of .25 MAC

300.20

3.0020

B.L. of .25 MAC

251.76

2.5176

Airfoil Section (Rockwell Mod NASA)

XXXX-64

Root  $\frac{b}{2}$  =

0.10

0.10

Tip  $\frac{b}{2}$  =

0.12

0.12

Data for (1) of (2) Sides

Leading Edge Cuff

Planform Area Ft<sup>2</sup>

118.333

0.01183

Leading Edge Intersects Fus M. L. @ Sta

500

5.000

Leading Edge Intersects Wing @ Sta

1083.4

10.834

**Notes:**

1. Positive directions of force coefficients, moment coefficients, and angles are indicated by arrow
2. For clarity, origins of wind and stability axes have been displaced from the center of gravity

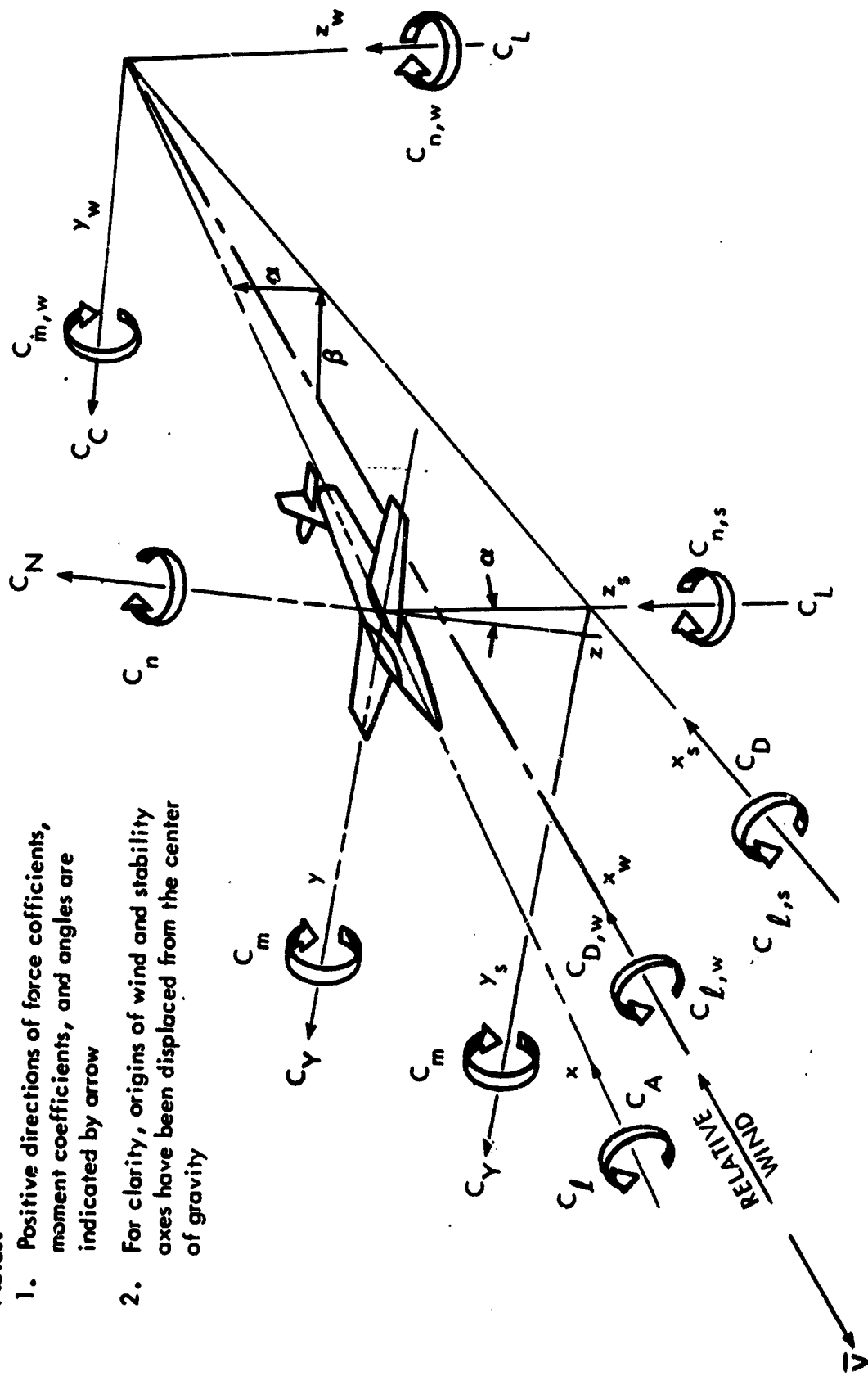


Figure 1. - Axis Systems.

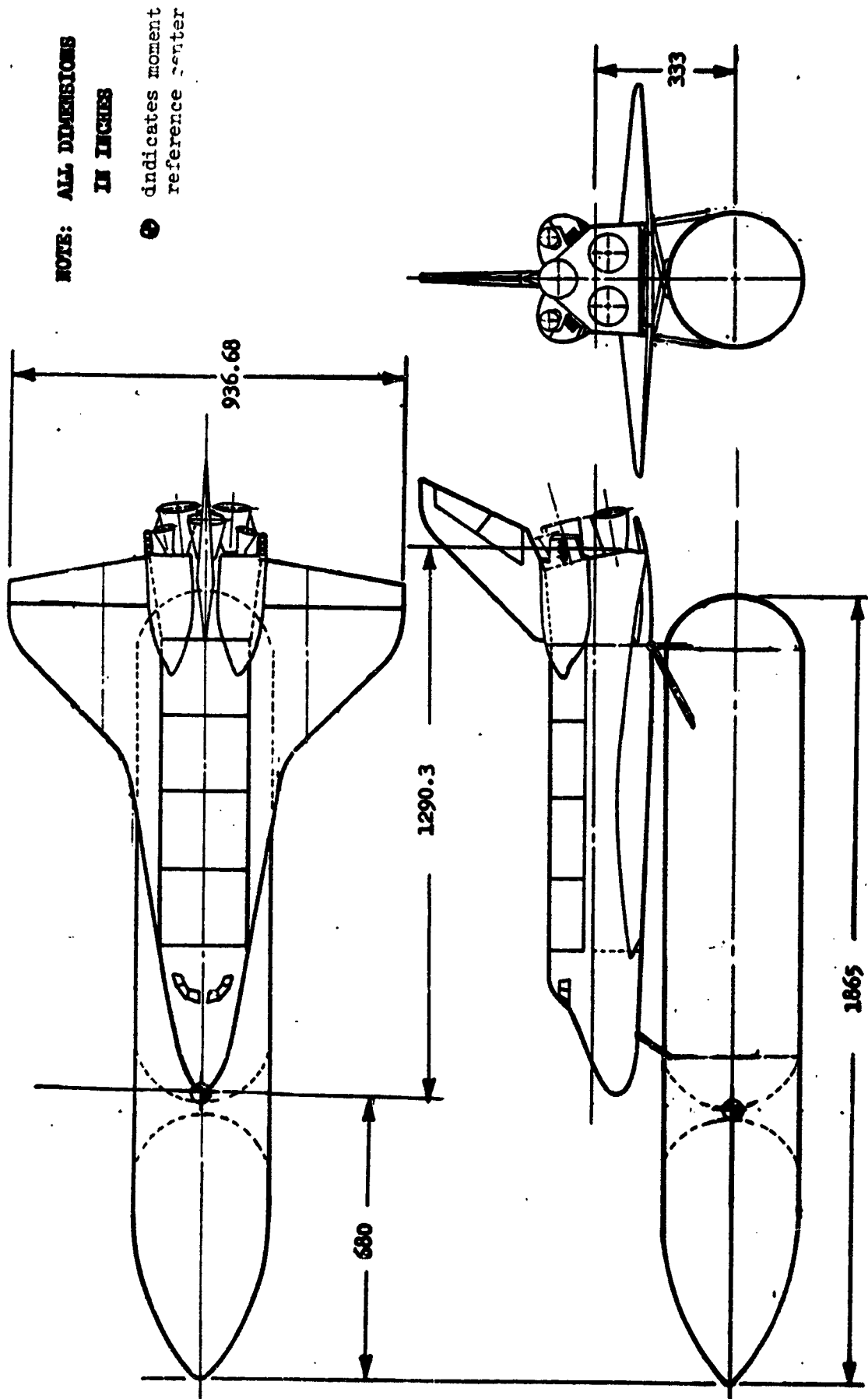
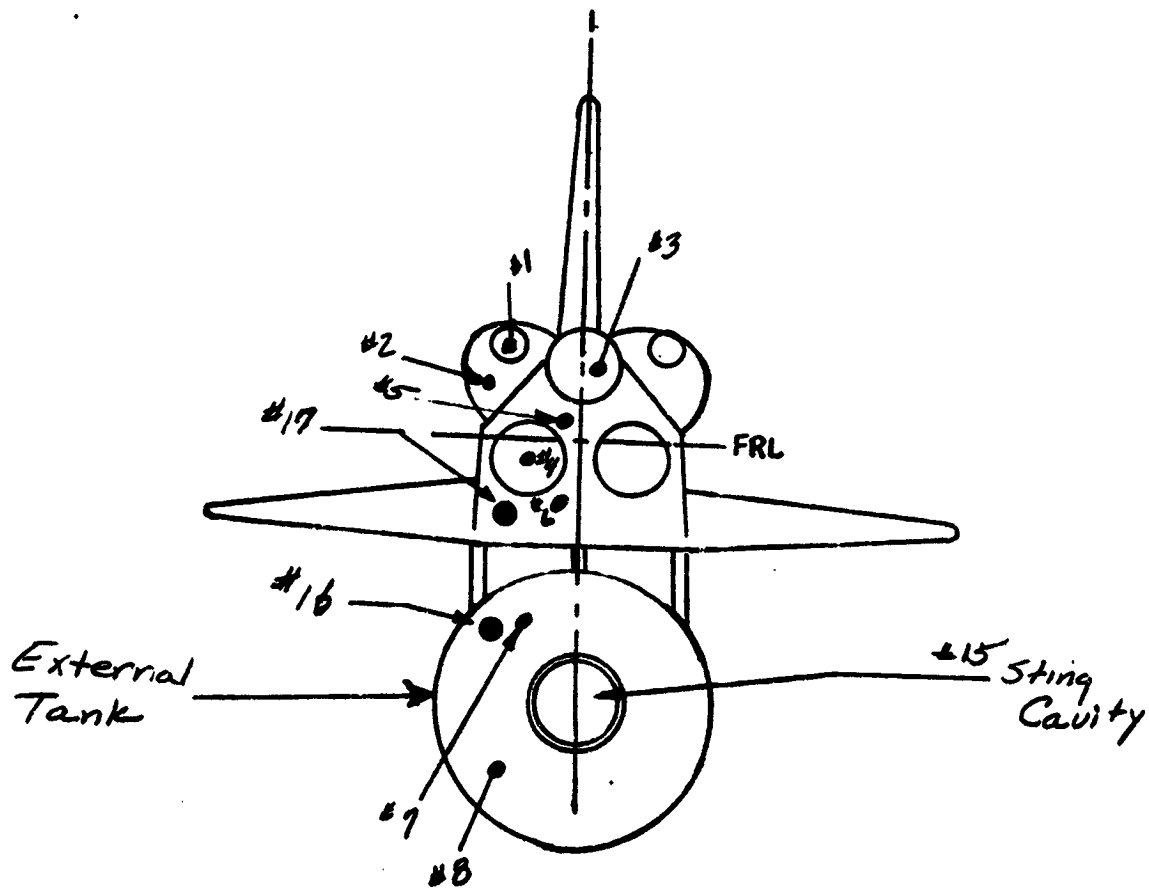


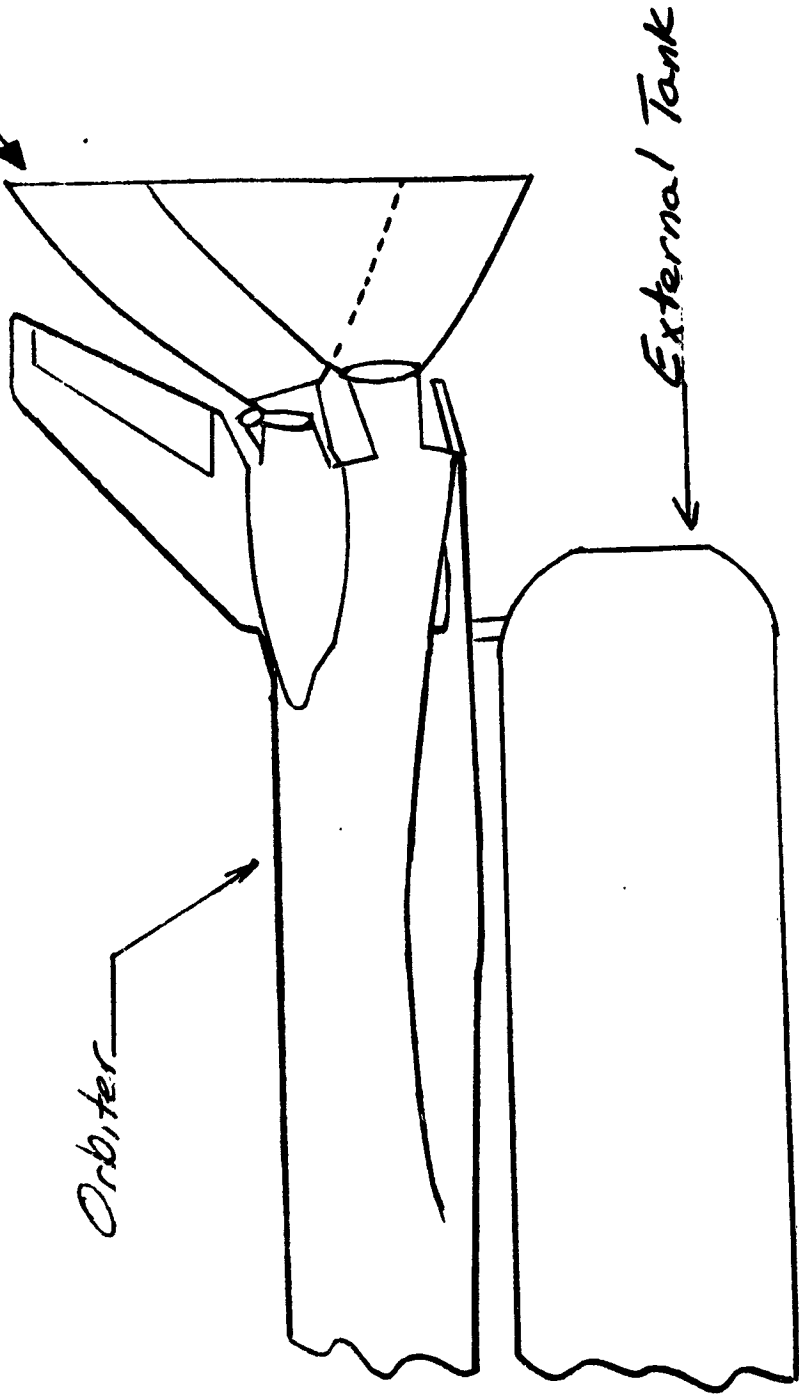
Figure 2. Model Sketches.  
a. Orbiter Tank General Arrangement, O<sub>9</sub>T<sub>10</sub>



b. Base Pressure Orifice Locations

Figure 2. - Continued.

*Non-Metric Plumes*



*Orbiter*

*External Tank*

C. Plume Simulators

Figure 2. - Concluded.



Figure 3. - Model installation photograph.



DATA FIGURES

DATA SET SYMBOL: (587010) (587011)

CONFIGURATION DESCRIPTION: 09 T10 AT2 PLUME ON; 09 T10 AT2 PLUME OFF

BETA: .000; AILTRON: .000; ELEVON: .000; RUDDER: .000

REFERENCE INFORMATION: SQ.FT. 2690.0000; IN. 1290.0000; LREF 1290.0000; IN. 936.6800; BREF 936.6800; IN. 1076.4800; XTRP 1076.4800; IN. 400.0000; YTRP 400.0000; IN. 400.0000; ZTRP 400.0000; IN. 400.0000; SCALE .0100

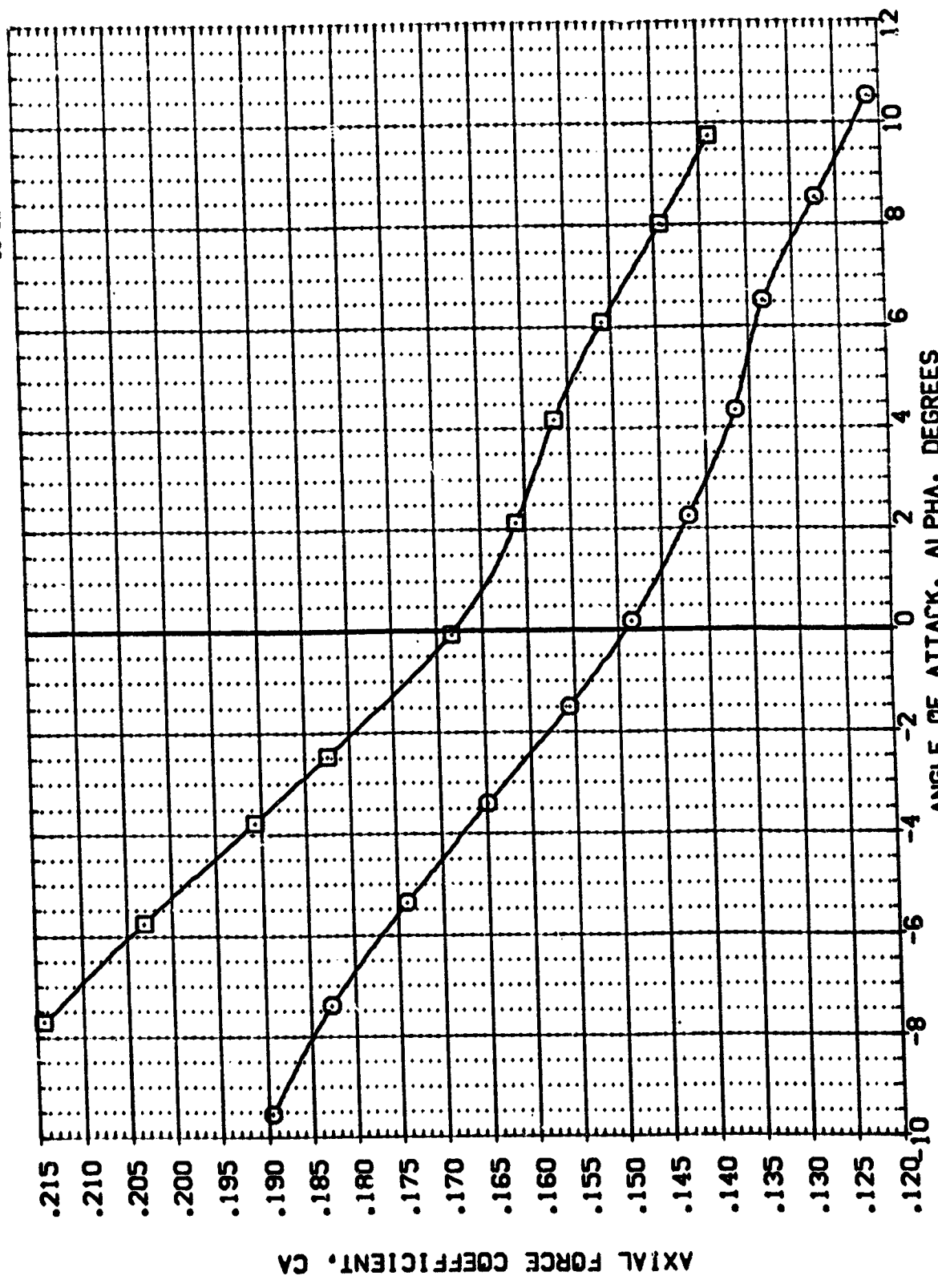


FIG. 4 EFFECTS OF SOLID PLOUMES ON LONGITUDINAL CHARACTERISTICS.

(A)MACH = 5.30

DATA SET SYMBL (S87010) (S87311)   
 CONFIGURATION DESCRIPTION AYES 3.5-169 IA10 OS T10 AT2 PLUME ON   
 AYES 3.5-169 IA10 OS T10 AT2 PLUME OFF

BETA .000 .000   
 AILRON .000 .000   
 ELEVEN .000 .000   
 RUDDER .000 .000

REFERENCE INFORMATION   
 SREF 2690.0000 SO.FT.   
 LREF 1290.0000 IN.   
 BREF 925.6800 IN.   
 XMRP 1076.4800 IN.   
 YMRP .0000 IN.   
 ZMRP 400.0000 IN.   
 SCALE 400.0100

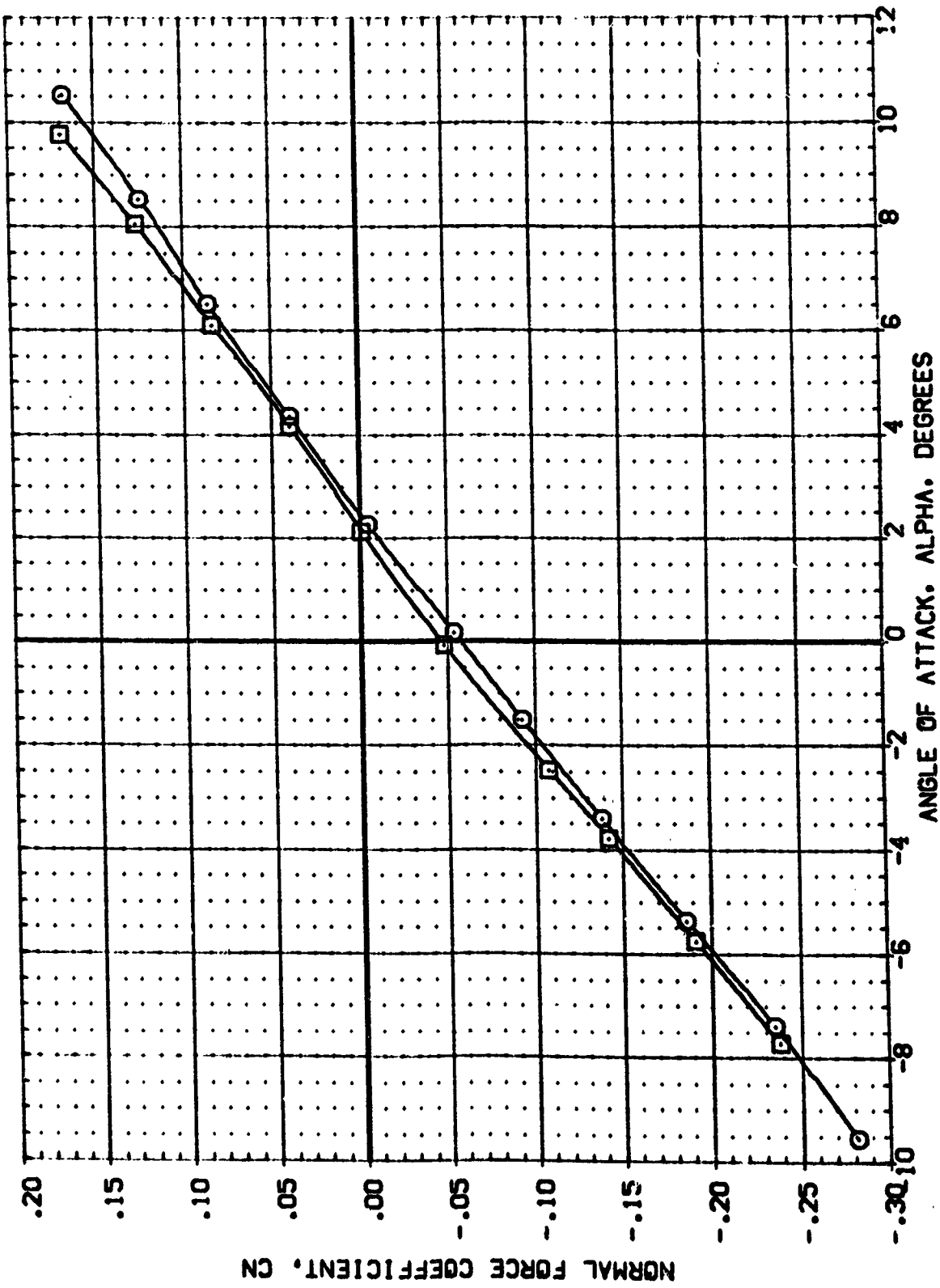


FIG. 4 EFFECTS OF SOLID PLOWES ON LONGITUDINAL CHARACTERISTICS.

(A)MACH = 5.30

DATA SET SYMB. CONFIGURATION DESCRIPTION  
 (587010) AVE 3.5-169 IA10 09 T10 AT2 PLUME ON  
 (087011) AVE 3.5-169 IA10 09 T10 AT2 PLUME OFF

BETA .000  
 AIRLON .000  
 ELEVON .000  
 RUDDER .000

REFERENCE INFORMATION  
 SREF 2690.0000 SO.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

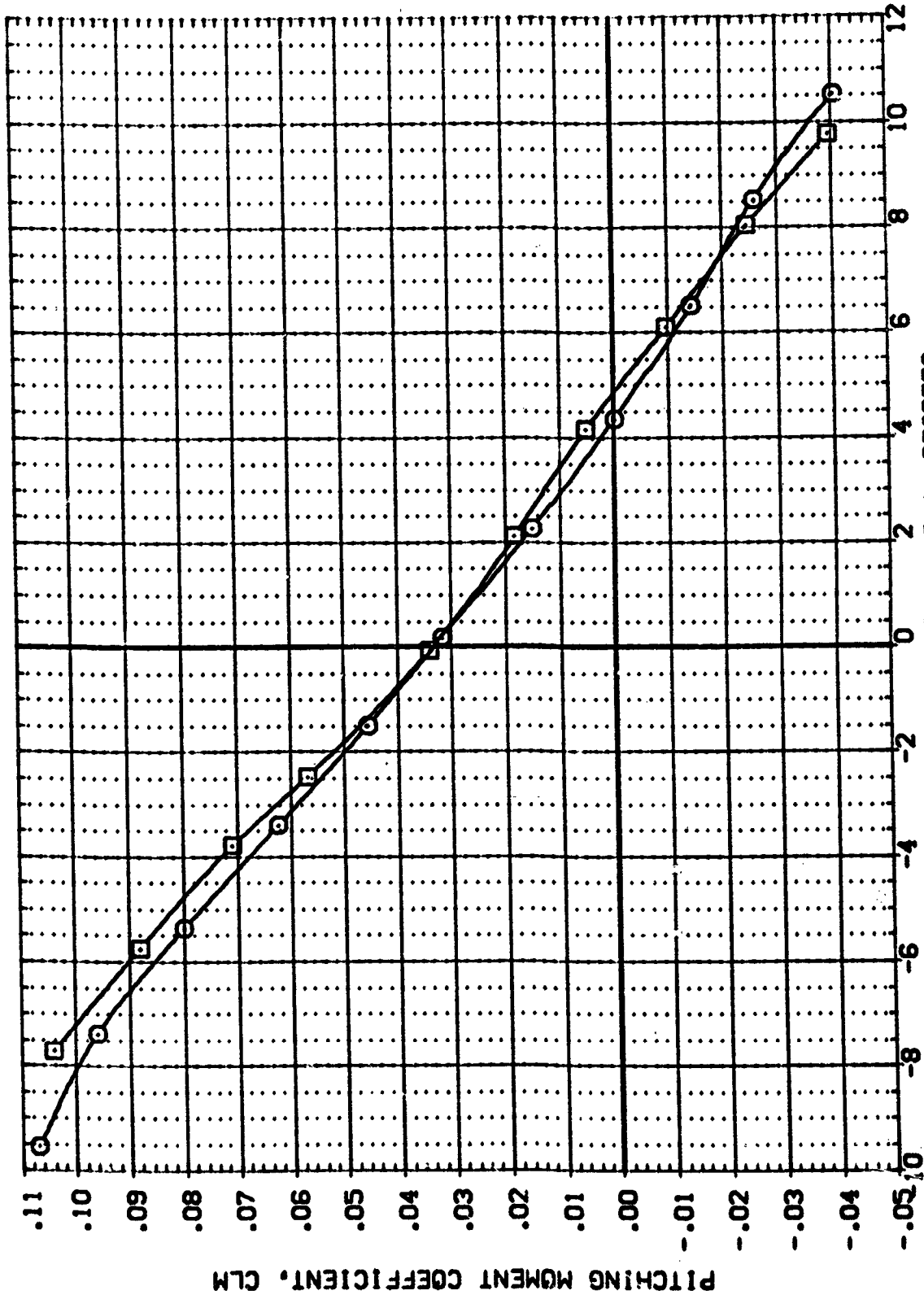


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.  
 (A)MACH = 5.30

DATA SET SYMBOL (SB7010) (DB7011) □

CONFIGURATION: DESCRIPTION

AVES 3.5-169 1A1/2 08 T10 AT2 PLUVE CN  
 AVES 3.5-169 1A1/0 08 T10 AT2 PLUVE CF

REFERENCE INFORMATION  
 SREF 2090.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

BETA .000  
 .000  
 AILRON .000  
 .000  
 ELEVON .000  
 .000  
 RUDDER .000  
 .000

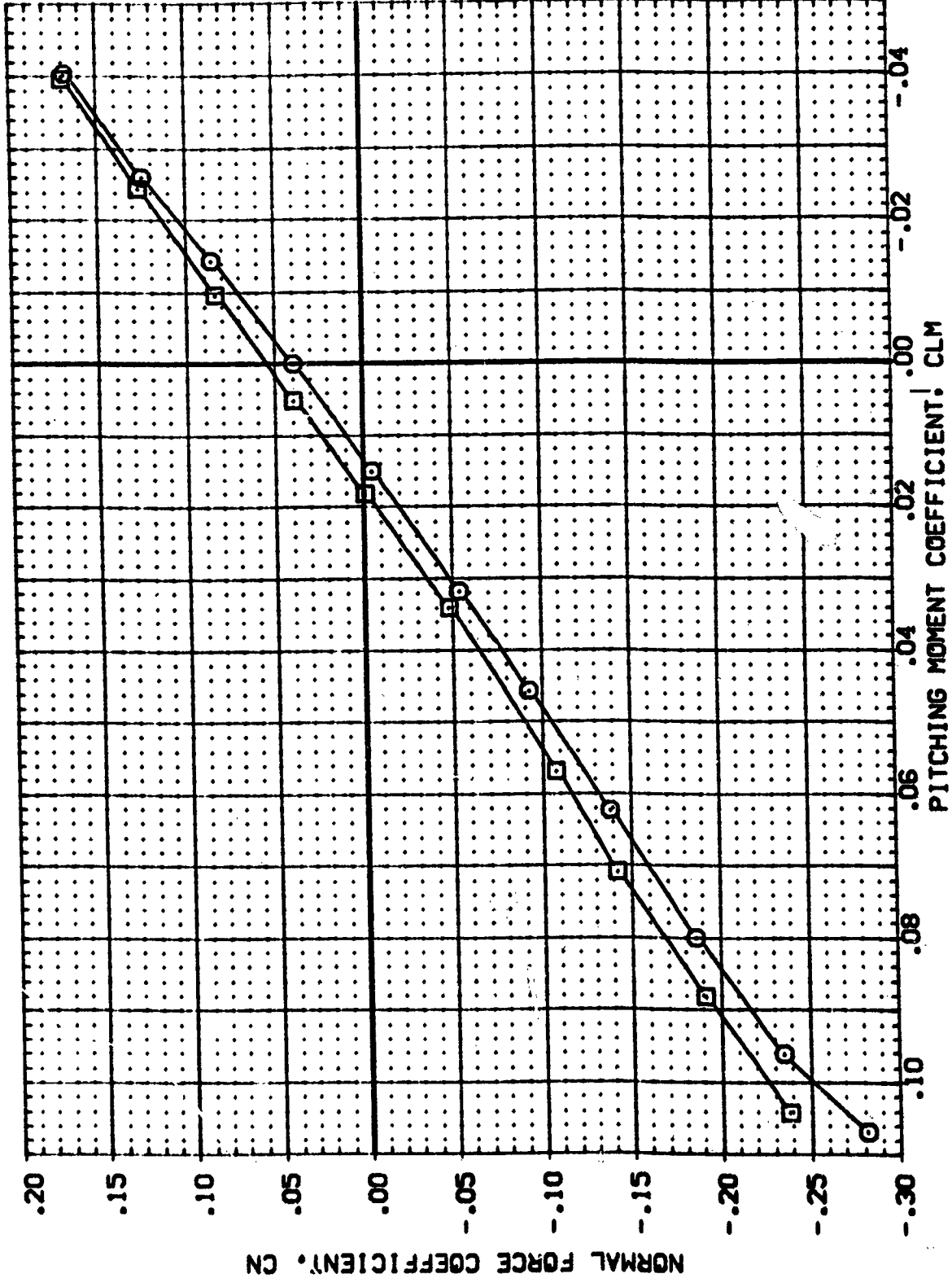


FIG. 4 EFFECTS OF SOLID PLOUMES ON LONGITUDINAL CHARACTERISTICS.

(M)MACH = 5.30



DATA SET SYMBOL (197010) (197001) □  
 CONFIGURATION DESCRIPTION  
 AVES 3.5-169 IA10 OS T10 AT2 PLUME ON  
 AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF

BETA .000 .000  
 AILTRON .000 .000  
 ELEVON .000 .000  
 RUDDER .000 .000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

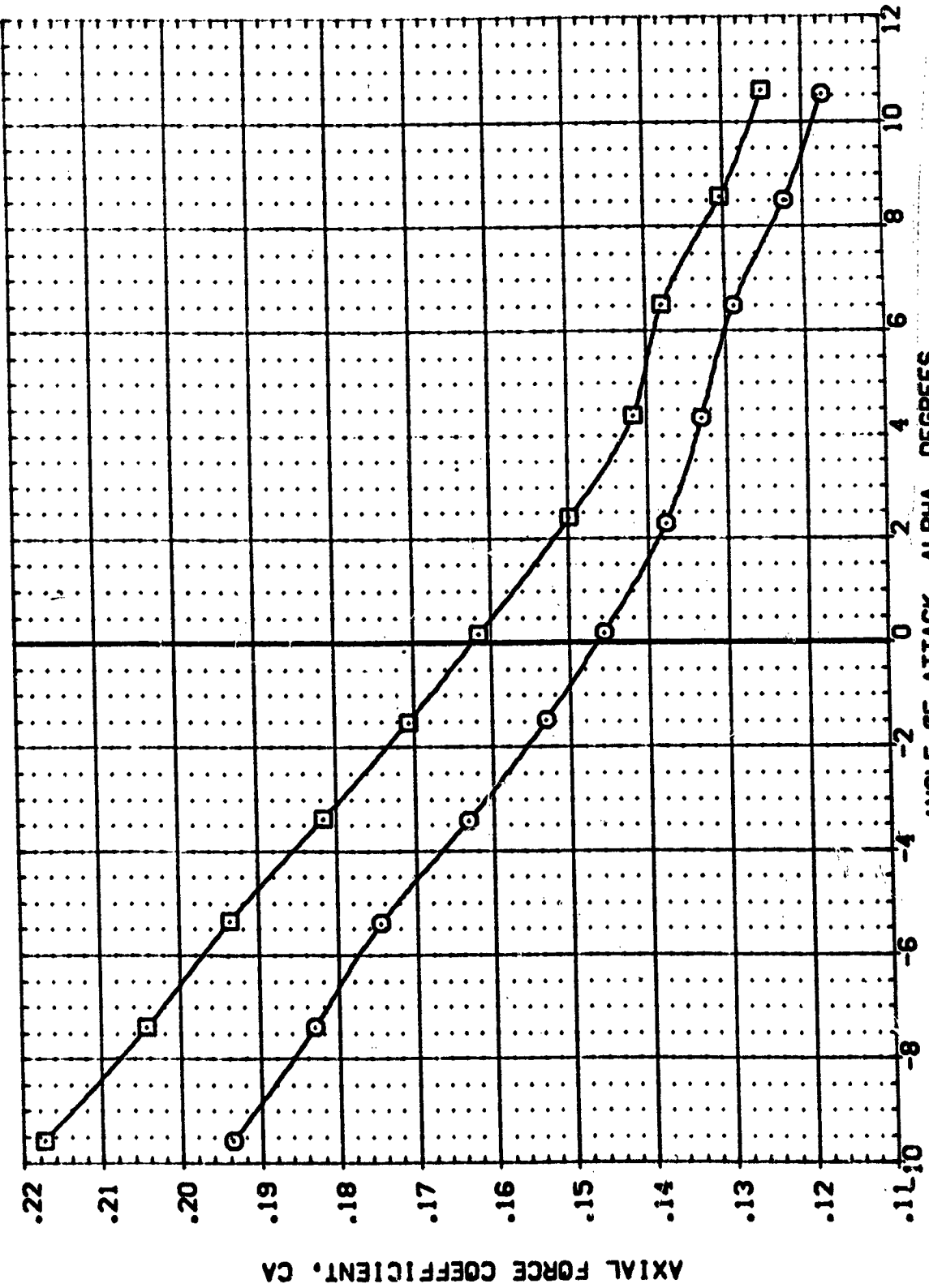


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.

(A)MACH = 7.32

DATA SET SYMBOL (TB7010) (RB7001) □

CONFIGURATION DESCRIPTION  
 APES 3.5-169 IA10 OS T10 AT2 PLUVE ON  
 APES 3.5-169 IA10 OS T10 AT2 PLUVE OFF

BETA .000 .000  
 AIRLON .000 .000  
 ELEVON .000 .000  
 RUDDER .000 .000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6600 IN.  
 XMRP 1076.4600 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

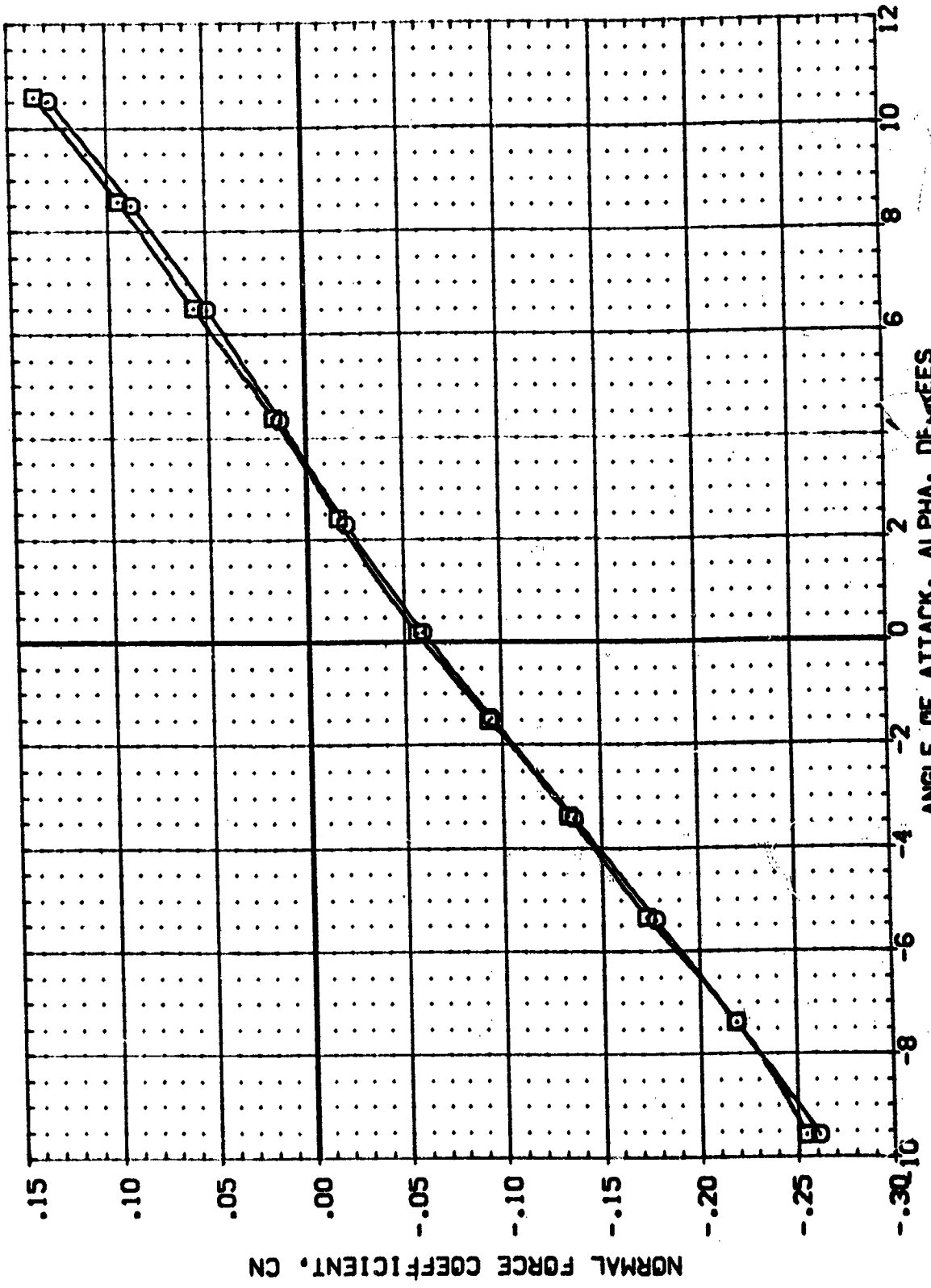


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.

(A)MACH = 7.32

DATA SET SYMBOL (1B7010) (REF7001)

CONFIGURATION DESCRIPTION  
 AVES 3.5-169 IA10 OS T10 AT2 PLUPE ON  
 AVES 3.5-169 IA10 OS T10 AT2 PLUPE OFF

BETA .000  
 AIRLON .000  
 ELEVON .000  
 RUDDER .000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 XMRP 936.6800 IN.  
 YMRP 1076.4800 IN.  
 ZMRP 400.0000 IN.  
 SCALE 400.0100

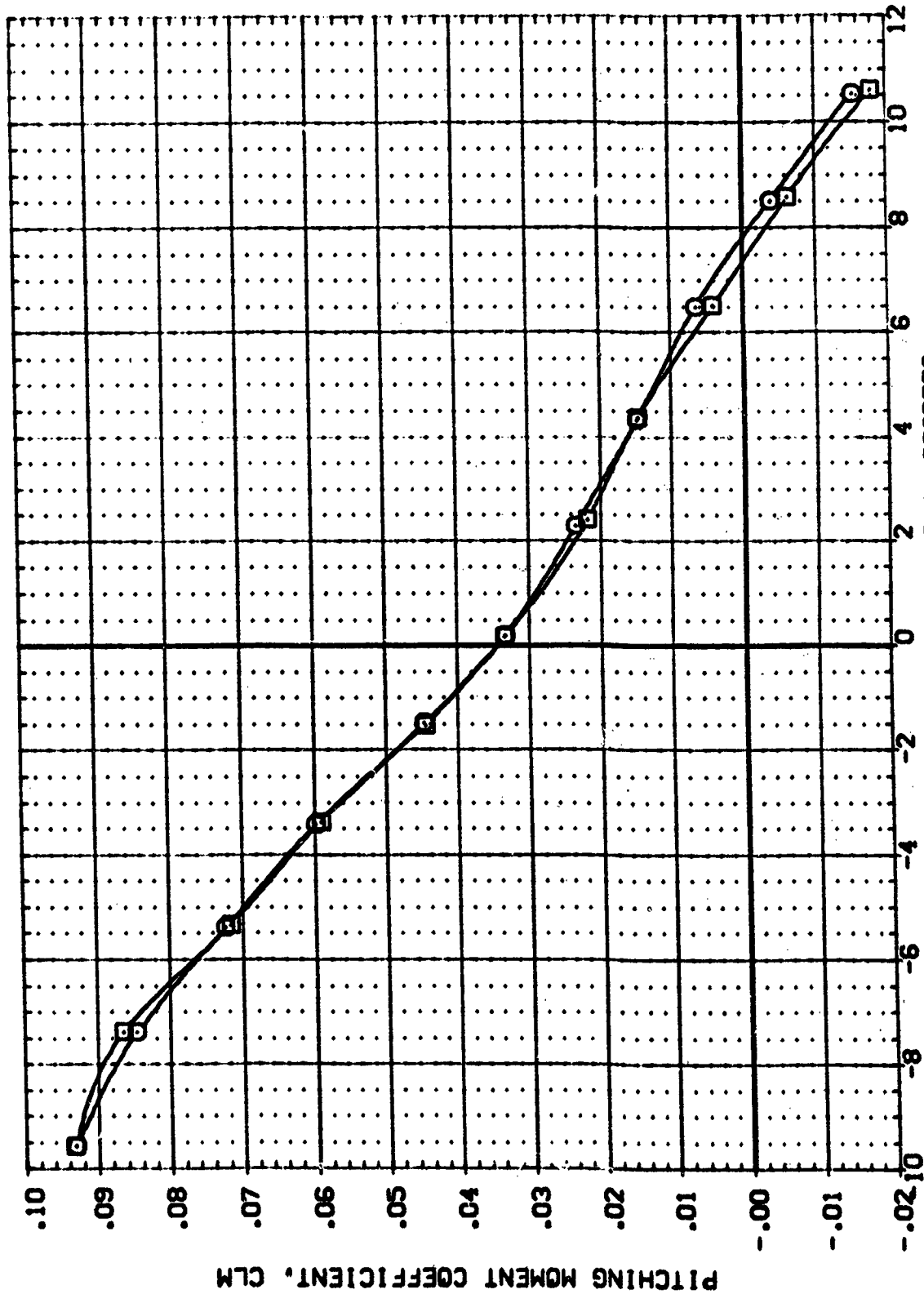


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.

(AJMACH = 7.32



DATA SET SYMBOL (R87010) (R87001)

CONFIGURATION DESCRIPTION  
 AVES 3.5-169 IA10 CS T10 AT2 PLUME ON  
 AVES 3.5-169 IA10 CS T10 AT2 PLUME OFF

BETA .000  
 AILIRON .000  
 ELEVON .000  
 RUDDER .000

REFERENCE INFORMATION  
 SREF 2690.0000 SO.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

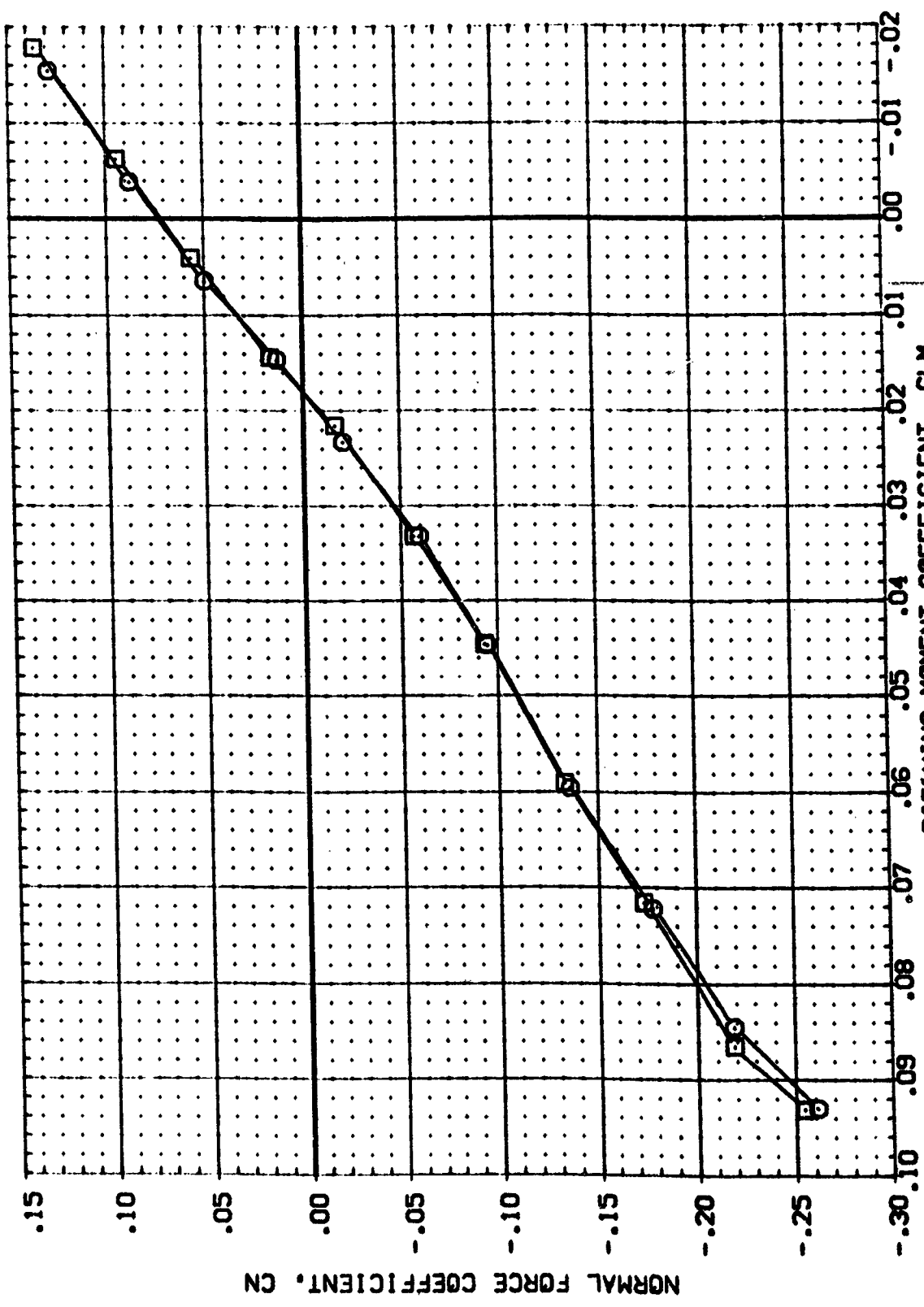


FIG. 4 EFFECTS OF SOLID PLOUMES ON LONGITUDINAL CHARACTERISTICS.

(AJMACH = 7.32



DATA SET SYMBOL: (R87009) (R87006)

CONFIGURATION DESCRIPTION: AVES 3.5-169 1A10 09 T10 AT2 PLUME ON / AVES 3.5-169 1A10 09 T10 AT2 PLUME OFF

ALPHA: .000 / .000

A1LRON: .000 / .000

ELEVON: .000 / .000

RUDER: .000 / .000

REFERENCE INFORMATION:

SREF	2690.0000	SO.FT.
LREF	1290.0000	IN.
BREF	976.6800	IN.
XTRP	1076.4800	IN.
YTRP	.0000	IN.
ZTRP	400.0000	IN.
SCALE	.0100	

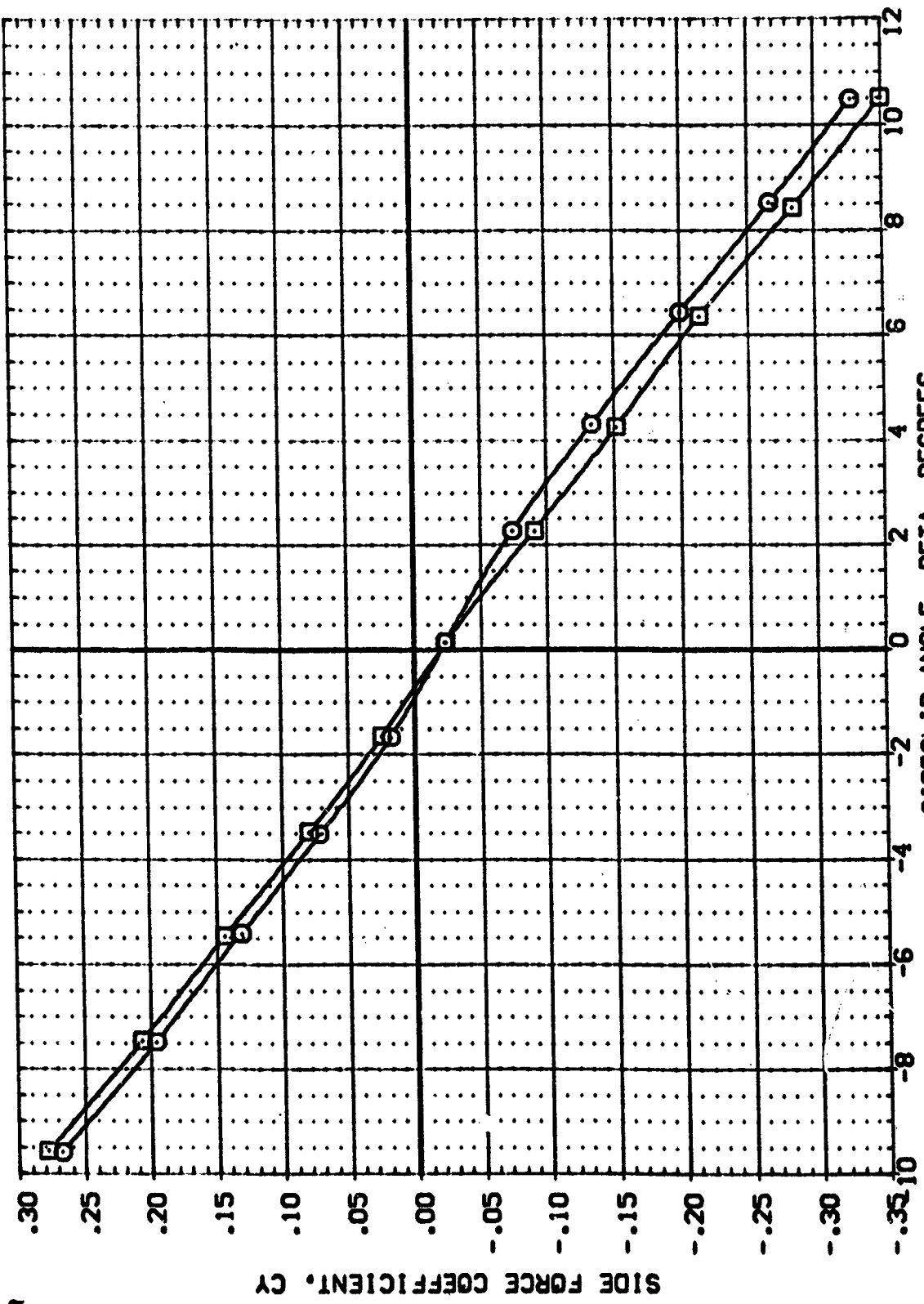


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMB. CONFIGURATION DESCRIPTION  
 (R87005) [ ] AVES 3.5-169 IA10 09 T10 AT2 PLUVE ON  
 (R87006) [ ] AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF

ALPHA AIRLON ELEVON RUDDER REFERENCE INFORMATION  
 .000 .000 .000 SREF 2690.0000 SQ.FT.  
 .000 .000 .000 LREF 1790.0000 IN.  
 .000 .000 .000 BREF 936.6800 IN.  
 .000 .000 .000 XMRP 1076.4800 IN.  
 .000 .000 .000 YMRP 400.0000 IN.  
 .000 .000 .000 ZMRP 400.0000 IN.  
 SCALE .0100

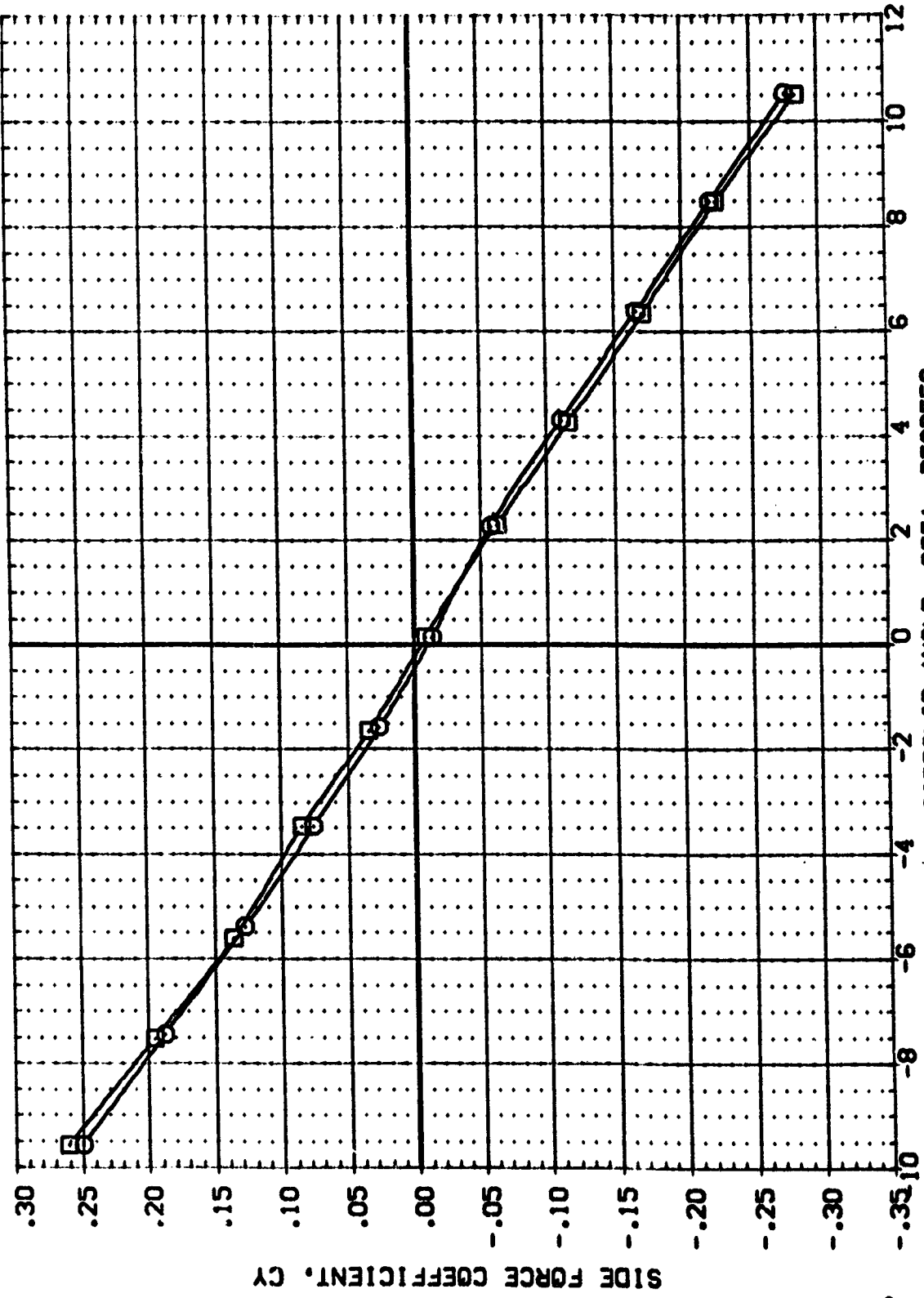


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (R87029) AVES 3.5-169 IA10 09 T10 AT2 PLUME ON  
 (R87035) AVES 3.5-169 IA10 09 T10 AT2 PLUME OFF

ALPHA .000 .000  
 AIRLON .000 .000  
 ELEVON .000 .000  
 RUDDER .000 .000  
 REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BRFP 936.6800 IN.  
 YPRP 1076.4800 IN.  
 ZPRP 400.0000 IN.  
 SCALE .0100

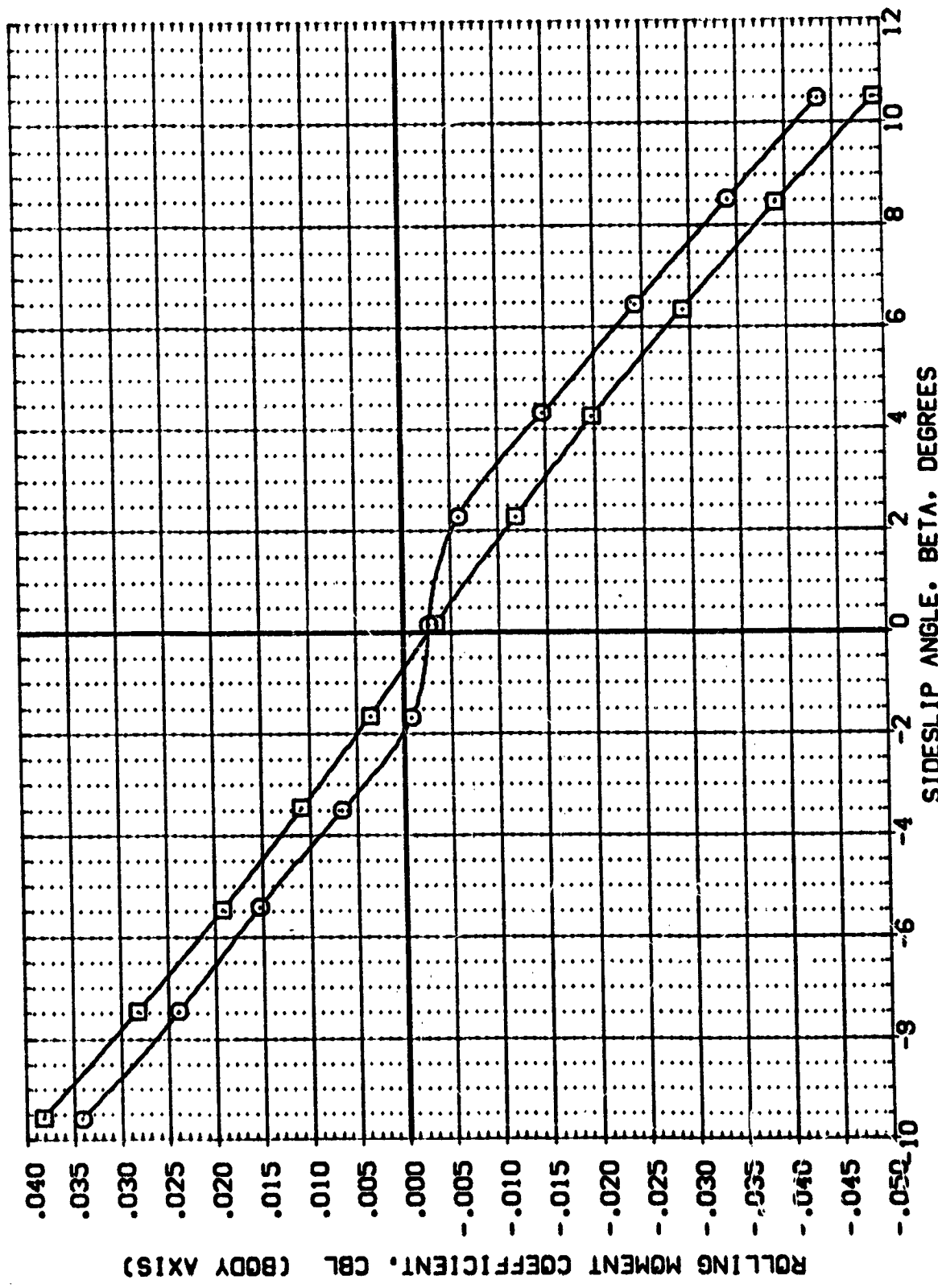


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AILERON	ELEVON	RUDDER	REFERENCE INFORMATION
(R87009)	AVES 3.5-169 IA10 OS T10 AT2 PLUME ON	.000	.000	.000	.000	SREF 2690.0000 SQ.FT.
(R87009)	AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF	.000	.000	.000	.000	LREF 1290.0000 IN.
						BREF 936.6800 IN.
						XMRP 1076.4800 IN.
						YMRP 400.0000 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

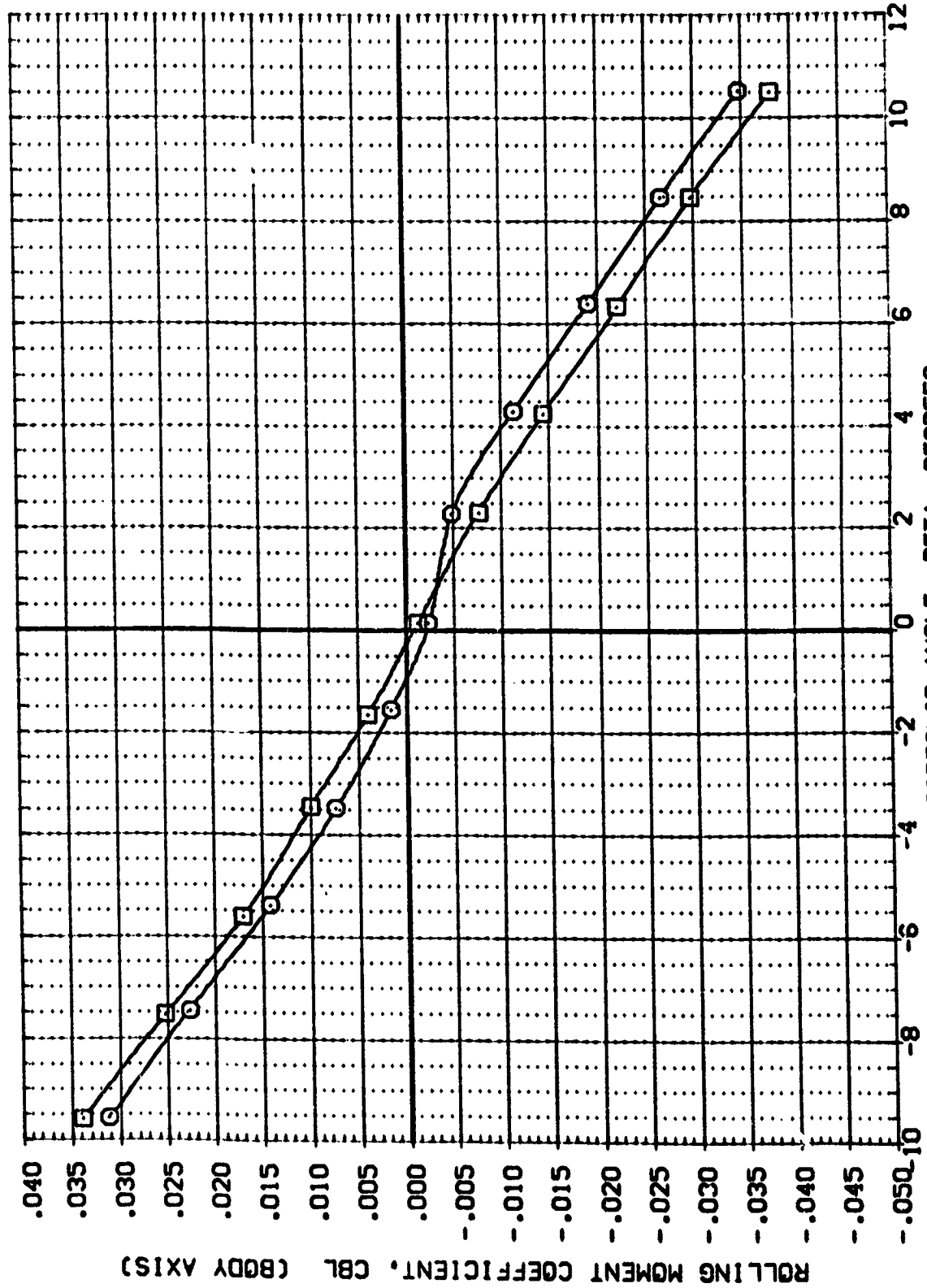


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL (R87009) (R87006) CONFIGURATION DESCRIPTION AMES 3.5-169 IA10 09 T10 AT2 PLUME ON AMES 3.5-169 IA10 09 T10 AT2 PLUME OFF

ALPHA .000 .000  
 AIRRDN .000 .000  
 ELEVON .000 .000  
 RUDDER .000 .000  
 REFERENCE INFORMATION SREF 2690.0000 SO.FT. LREF 1290.0000 IN. BREF 936.6800 IN. XMRP 1076.4800 IN. YMRP .0000 IN. ZMRP 400.0000 IN. SCALE .0100

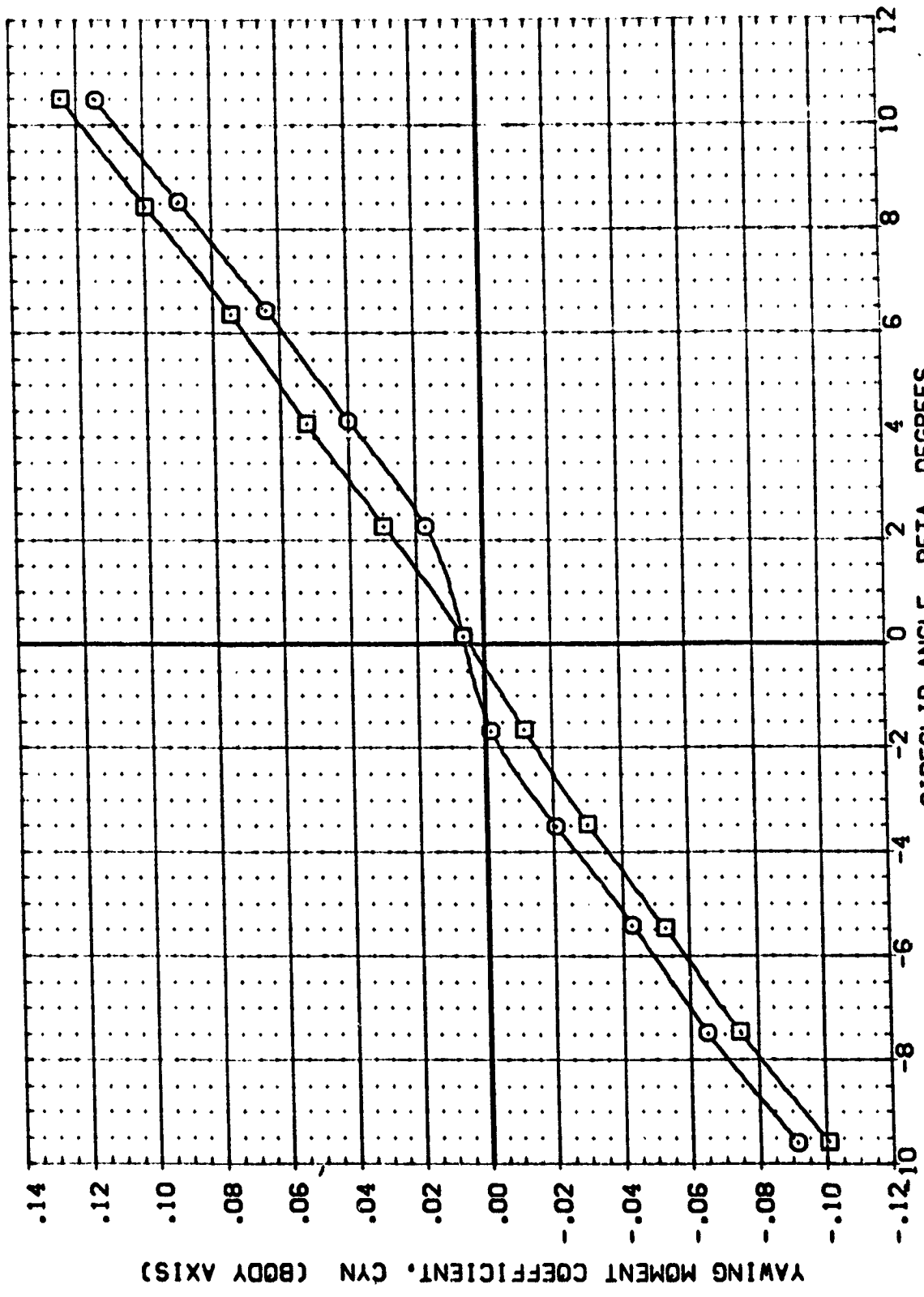


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AILRON	ELEVON	RUDDER	REFERENCE INFORMATION
(RB70C9)	AMES 3.5-169 IA10 09 T10 AT2 PLUFE ON	.000	.000	.000	.000	SREF 2690.0000 SQ.FT.
(RB70C6)	AMES 3.5-169 IA10 09 T10 AT2 PLUFE OFF	.000	.000	.000	.000	LREF 1290.0000 IN.
						BREF 936.6800 IN.
						XMRP 1076.4800 IN.
						YMRP .0000 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

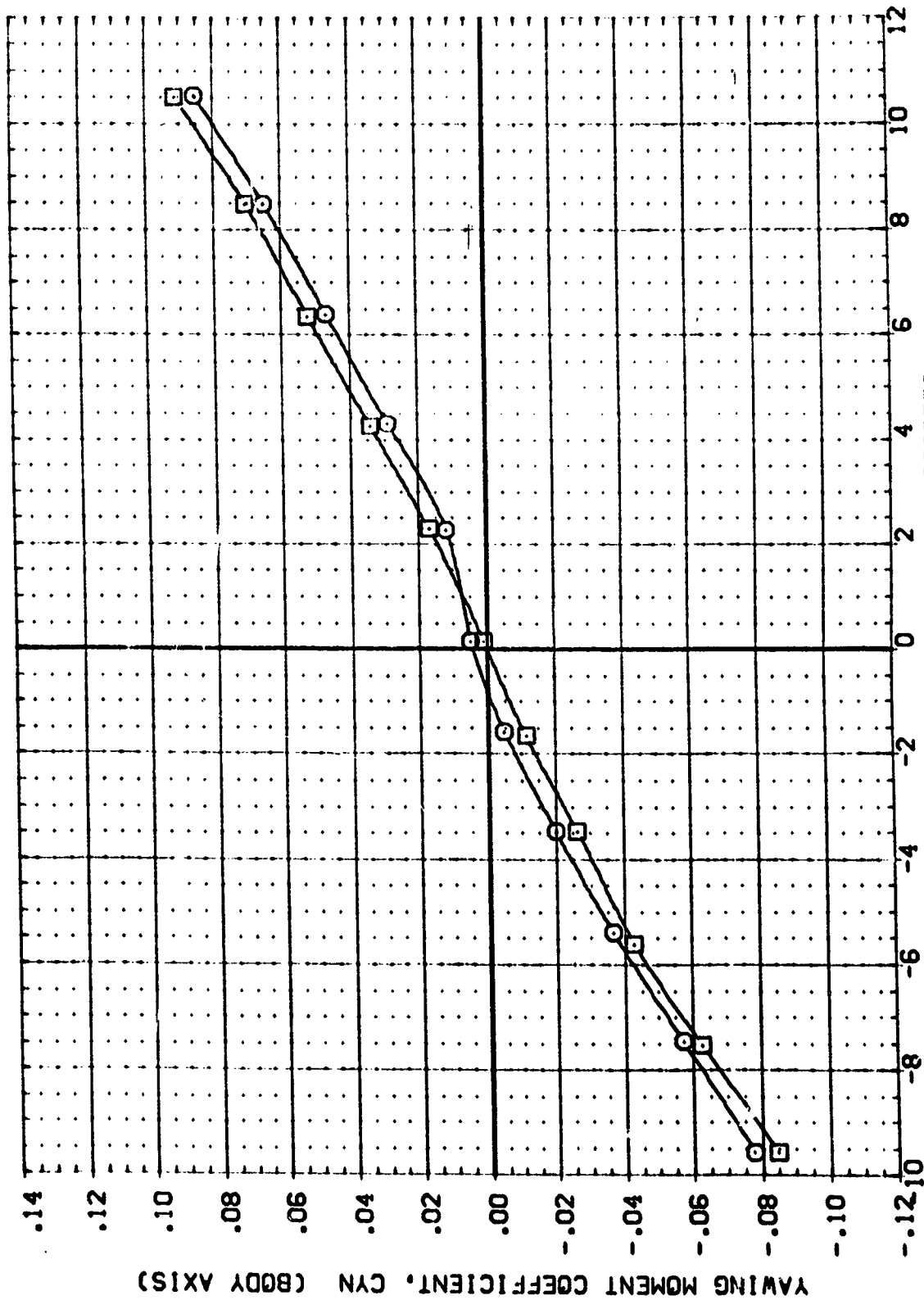


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.  
 (B)MACH = 7.32

DATA SET SYMBOL (RB7009) (RB7006)

CONFIGURATION DESCRIPTION  
 AVES 3.5-169 IA10 OS T10 AT2 PLUME ON  
 AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF

ALPHA .000 .000  
 AILRON .000 .000  
 ELEVON .000 .000  
 RUDDER .000 .000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

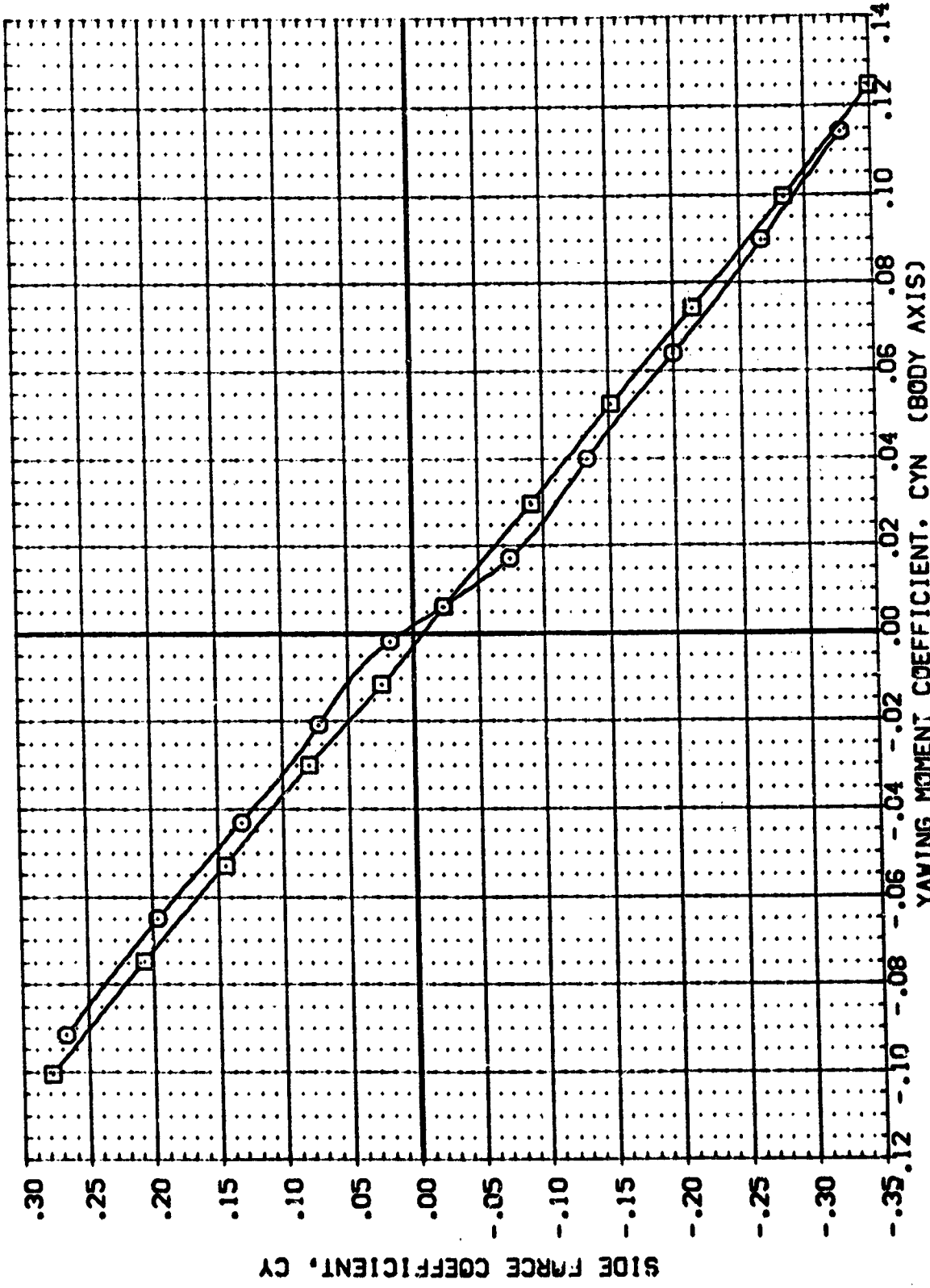


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26



DATA-SET SYMBOL: (R87009) (R87006)

CONFIGURATION DESCRIPTION:  
 AVES 3.5-169 IA10 09 T10 AT2 PLUVE ON  
 AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF

REFERENCE INFORMATION:  
 SREF 2690.0000 SO.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XTRP 1076.4800 IN.  
 YTRP .0000 IN.  
 ZTRP .0000 IN.  
 SCALE 400.0000 IN.  
 .0100

ALPHA .000  
 AIRWON .000  
 ELEVON .000  
 RUDDER .000

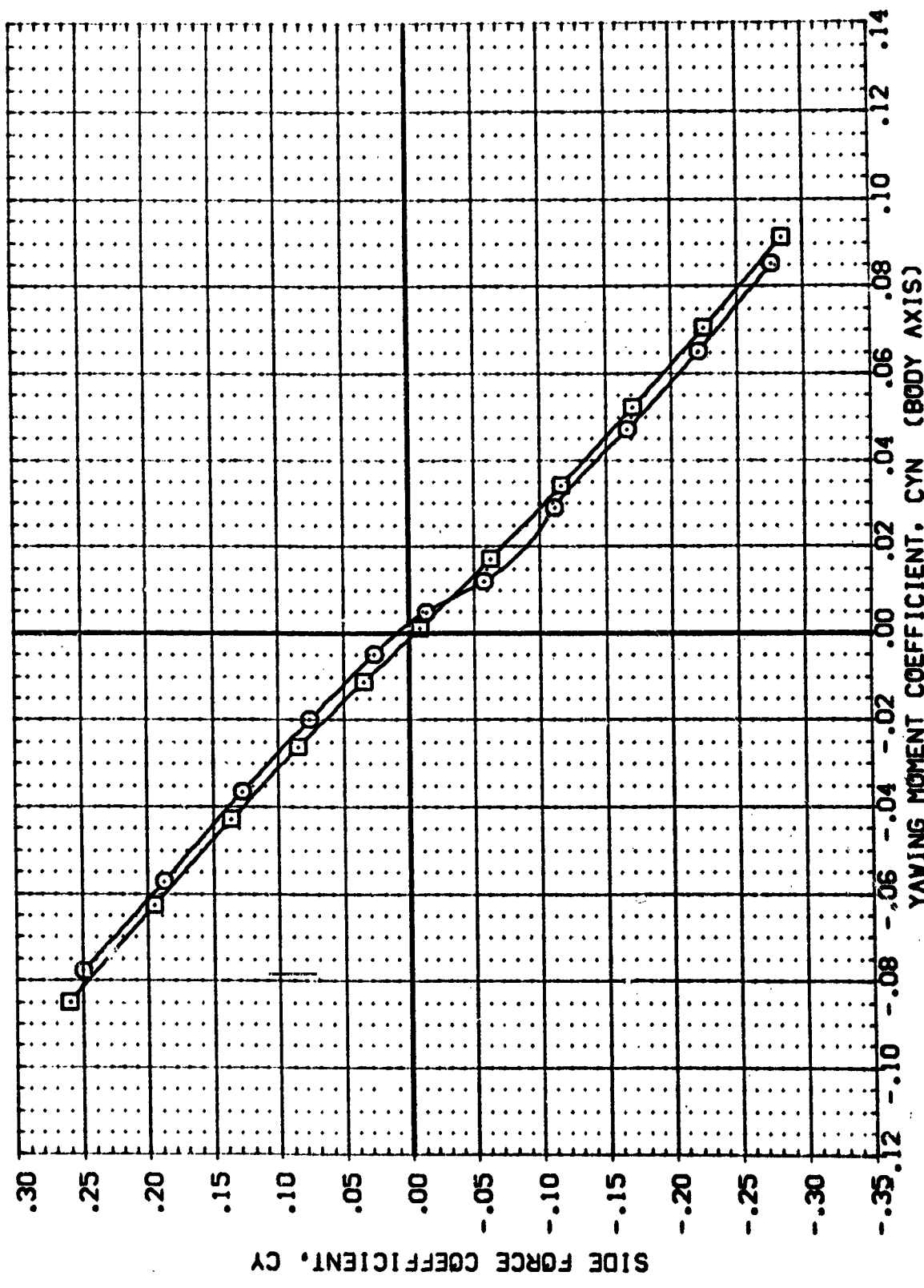


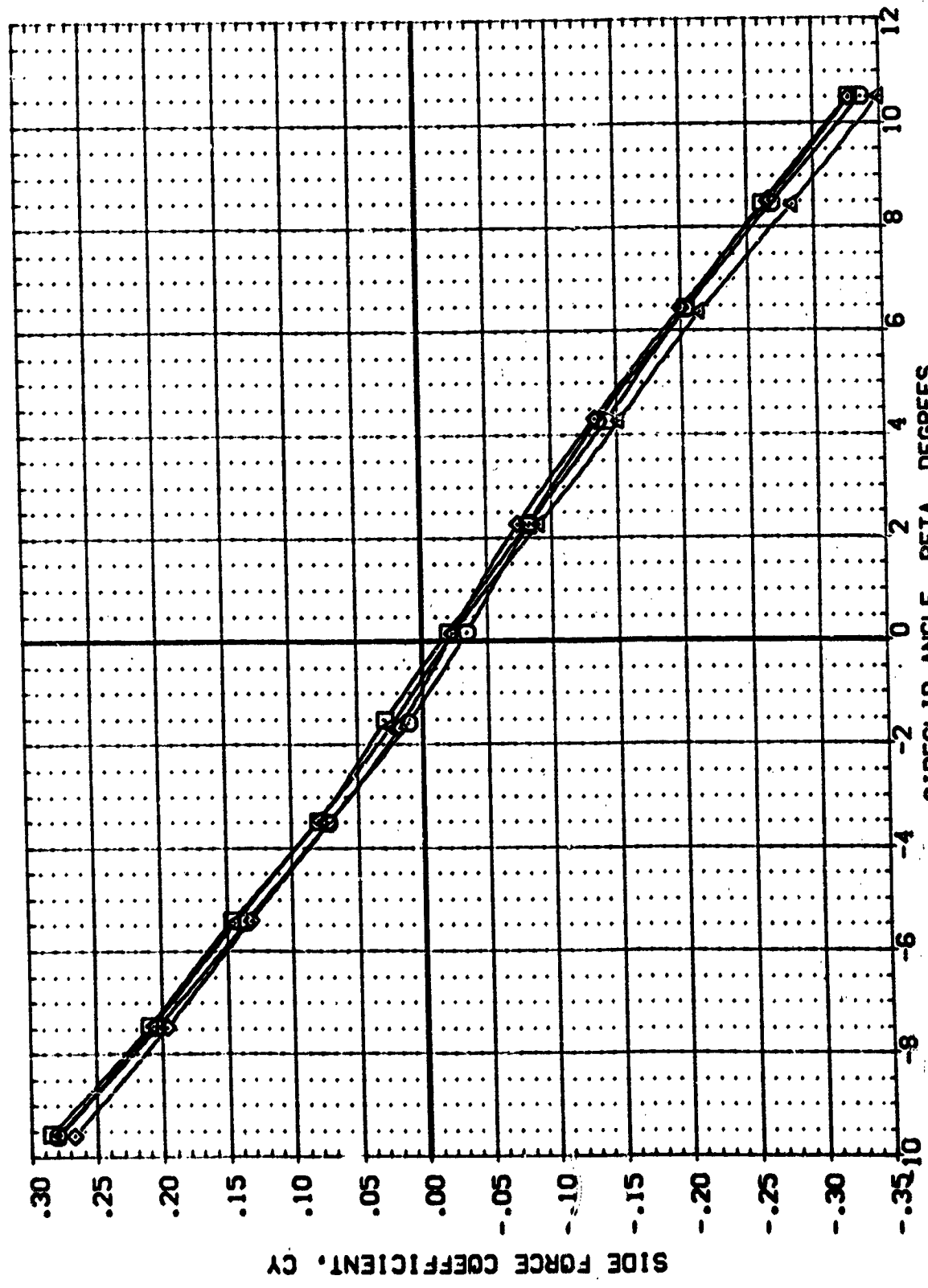
FIG. 5 EFFECTS OF SOLID PLOUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE 400.0100

RUDDER 10.000  
 ELEVON .000  
 AILERON .000  
 ALPHA .000  
 .000  
 .000  
 .000

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (R87008) AMES 3.5-169 IA10 09 T10 AT2 PLUME ON  
 (R87007) AMES 3.5-169 IA10 09 T10 AT2 PLUME OFF  
 (R87009) AMES 3.5-169 IA10 09 T10 AT2 PLUME ON  
 (R87006) AMES 3.5-169 IA10 09 T10 AT2 PLUME OFF



SIDESLIP ANGLE, BETA, DEGREES

FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(A)MACH = 5.26

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AILTRON	ELEVON	RUDDER	REFERENCE INFORMATION
(RBT008)	AVES 3.5-169 IA10 OS T10 AT2 PLUME ON	.000	.000	.000	10.000	SREF 2690.0000 SQ.FT.
(RBT007)	AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
(RBT009)	AVES 3.5-169 IA10 OS T10 AT2 PLUME ON	.000	.000	.000	.000	BREF 936.6800 IN.
(RBT006)	AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF	.000	.000	.000	.000	XMRP 1076.4800 IN.
						ZMRP .0000 IN.
						SCALE 400.0000
						.0100

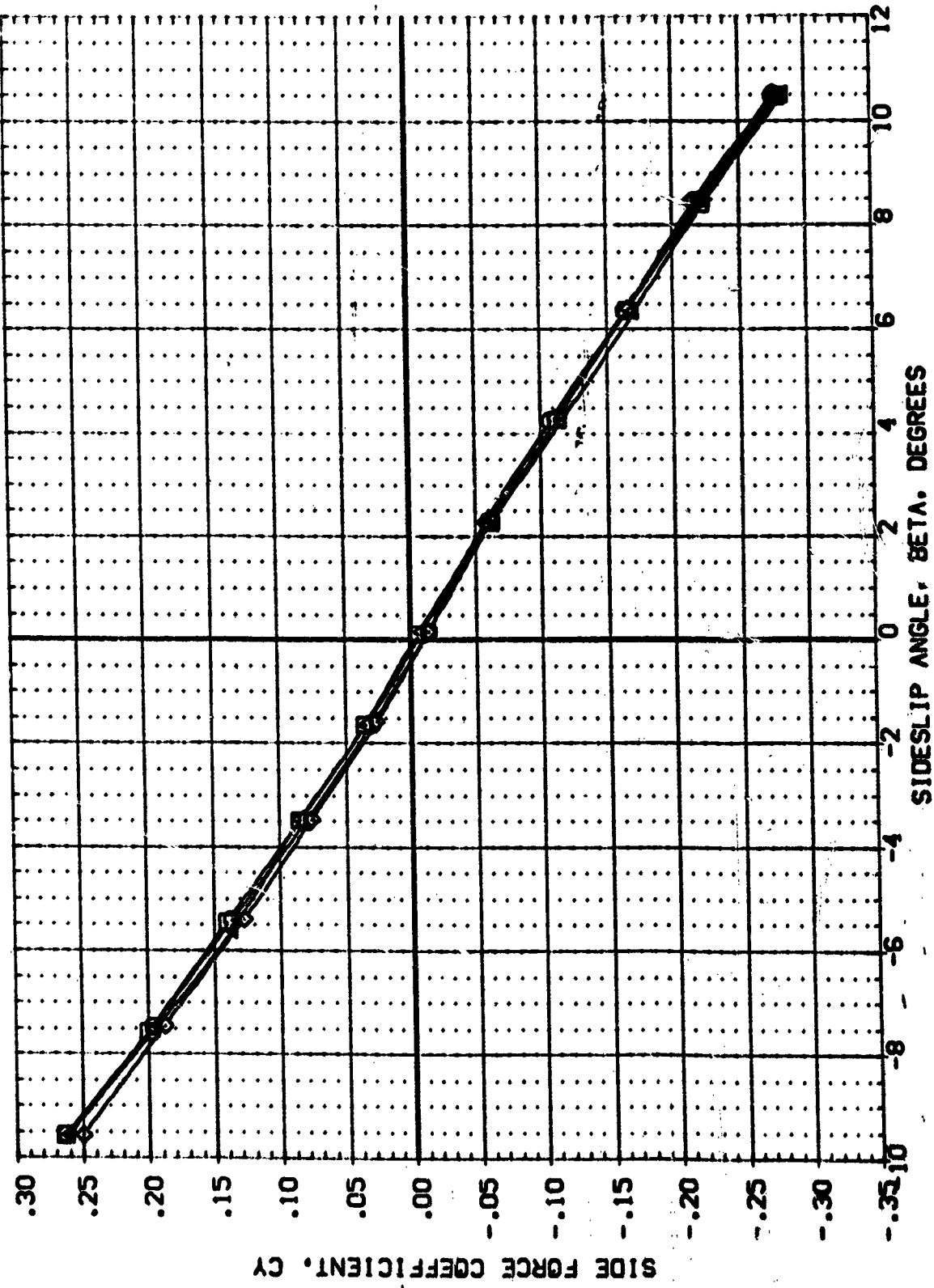


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(8)MACH = 7.32

DATA SET SYMBL	CONFIGURATION DESCRIPTION	ALPHA	AT10	ELEVEN	RUDDER	REFERENCE INFORMATION
(R87008)	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	10.000	SREF 2690.0000 SQ.FT.
(R87007)	AVES 3.5-169 IA10 09 T10 AT2 PLUME OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
(R87009)	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	.000	BREF 936.6800 IN.
(R87006)	AVES 3.5-169 IA10 09 T10 AT2 PLUME OFF	.000	.000	.000	.000	XPRP 1076.4800 IN.
						YPRP .0000 IN.
						ZPRP .0000 IN.
						SCALE .0100

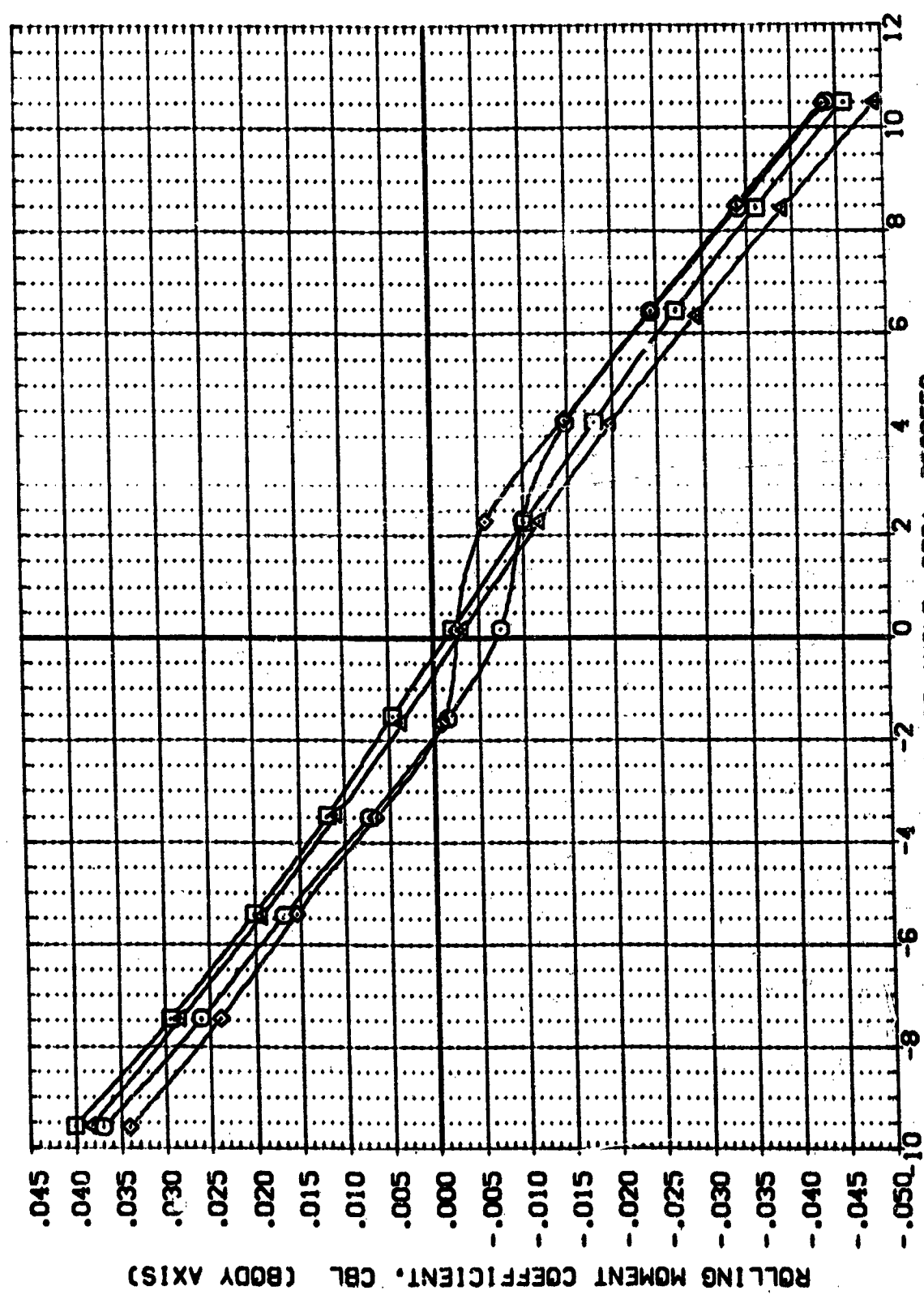


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(A)MACH = 5.26

DATA SET SYMBOL    CONFIGURATION DESCRIPTION  
 (R87008)    AVES 3.5-169 IA10 OS T10 AT2 PLUME ON  
 (R87007)    AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF  
 (R87006)    AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF

ALPHA    .000  
 AIRLON    .000  
 ELEVON    .000  
 RUDDER    10.000  
 REFERENCE INFORMATION  
 SREF    2690.0000    SQ.FT.  
 LREF    1290.0000    IN.  
 BREF    936.6800    IN.  
 XMRP    1076.4800    IN.  
 YMRP    .0000    IN.  
 ZMRP    400.0000    IN.  
 SCALE    .0100

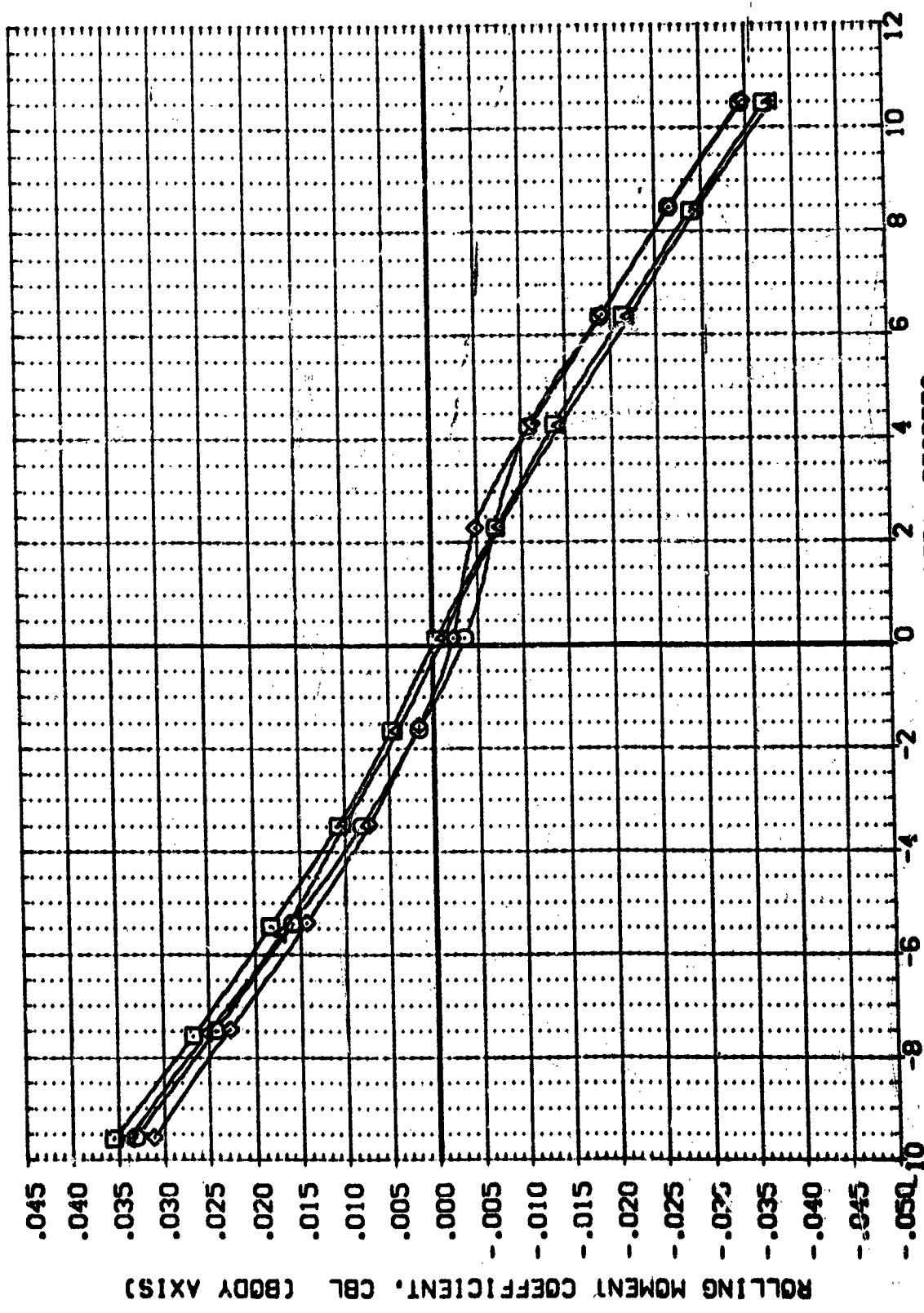


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(B)MACH = 7.32

DATA SET SYMBOL: (R67008), (R67007), (R67009), (R67006)

CONFIGURATION DESCRIPTION: AVES 3.5-169 IA1D, AVES 3.5-169 IA1B, AVES 3.5-169 IA1C

REFERENCE INFORMATION: SREF 2690.0000 SO.FT., LREF 1290.0000 IN., EREF 936.6800 IN., YPRP 1076.4800 IN., ZPRP 400.0000 IN., SCALE .0100

ALPHA: .000, .000, .000, .000

AILERON: .000, .000, .000, .000

ELEVON: .000, .000, .000, .000

RUDDER: 10.000, 10.000, .000, .000

AT2 PLUME ON: CS T10, CS T10, CS T10, CS T10

AT2 PLUME OFF: CS T10, CS T10, CS T10, CS T10

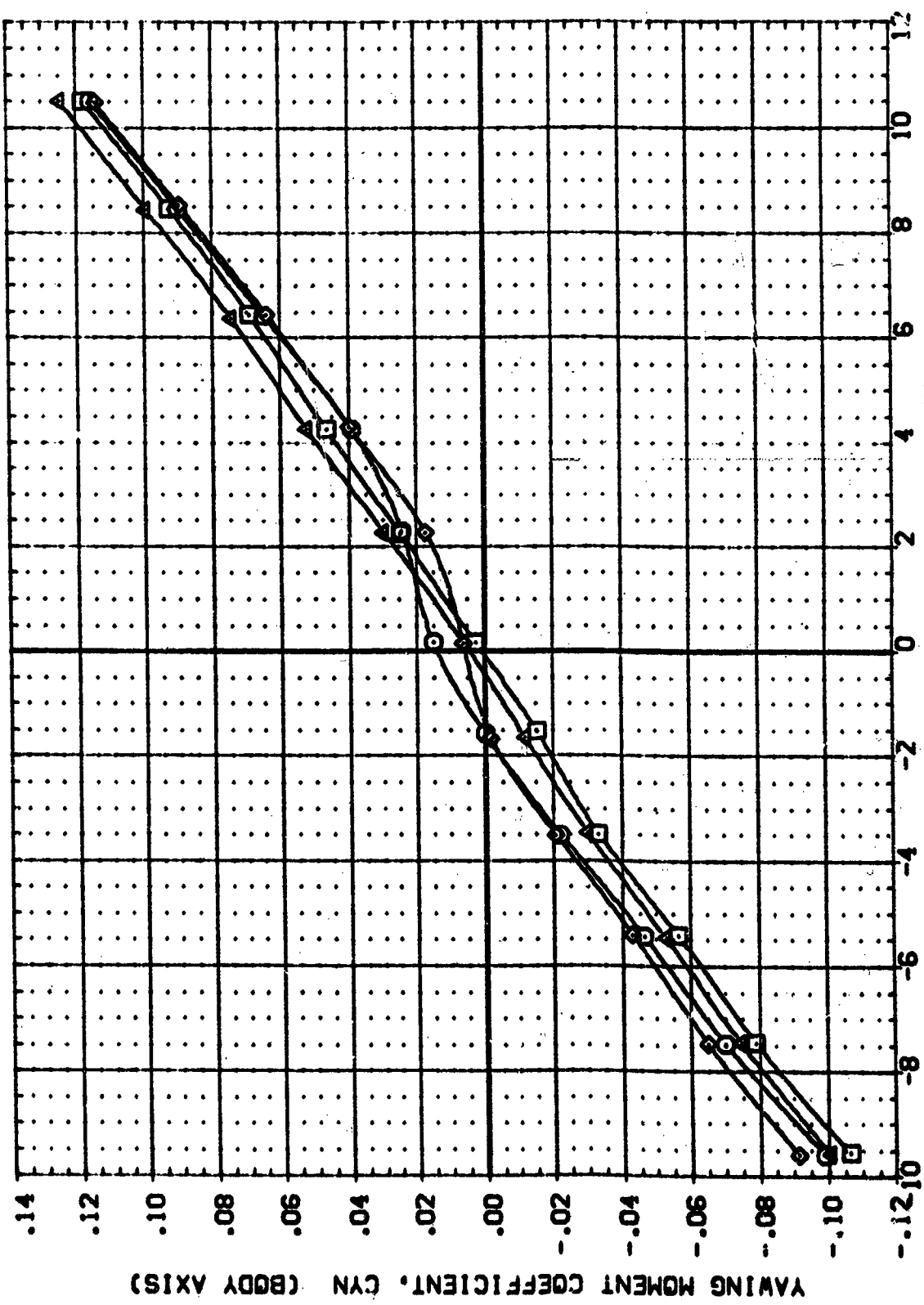


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(M)MACH = 5.26

DATA SET SYMBOL: (RB7008) (RB7007) (RB7009) (RB7006)

CONFIGURATION DESCRIPTION: AVES 3.5-169 IA10 OS T10 AT2 PLUME ON (RB7008) AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF (RB7007) AVES 3.5-169 IA10 OS T10 AT2 PLUME ON (RB7009) AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF (RB7006)

ALPHA: .000 .000 .000 .000

AIRLON: .000 .000 .000 .000

ELEVON: .000 .000 .000 .000

RUDDER: 10.000 10.000 .000 .000

REFERENCE INFORMATION: SREF 2690.0000 SQ.FT. LREF 1290.0000 IN. BREF 936.6800 IN. XMRP 1076.4800 IN. YMRP 400.0000 IN. SCALE .0100

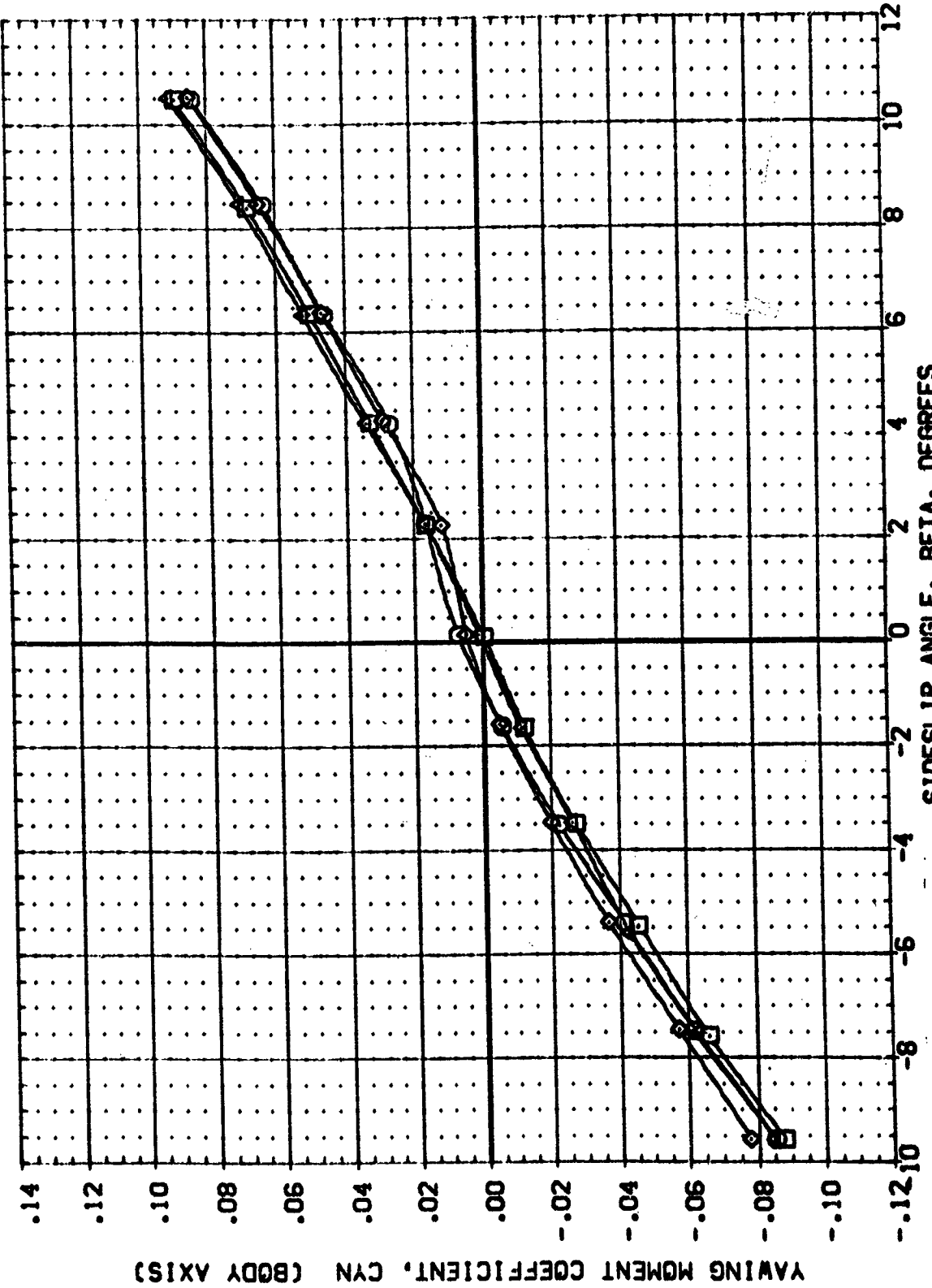


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(B)MACH = 7.32

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AILERON	ELEVON	RUDDER	REFERENCE INFORMATION
(R87008)	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	10.000	SREF 2690.0000 SQ.FT.
(R87007)	AVES 3.5-169 IA10 09 T10 AT2 PLUME OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
(R87009)	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	.000	BREF 936.6800 IN.
(R87005)	AVES 3.5-169 IA10 09 T10 AT2 PLUME OFF	.000	.000	.000	.000	XMRP 1076.4800 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

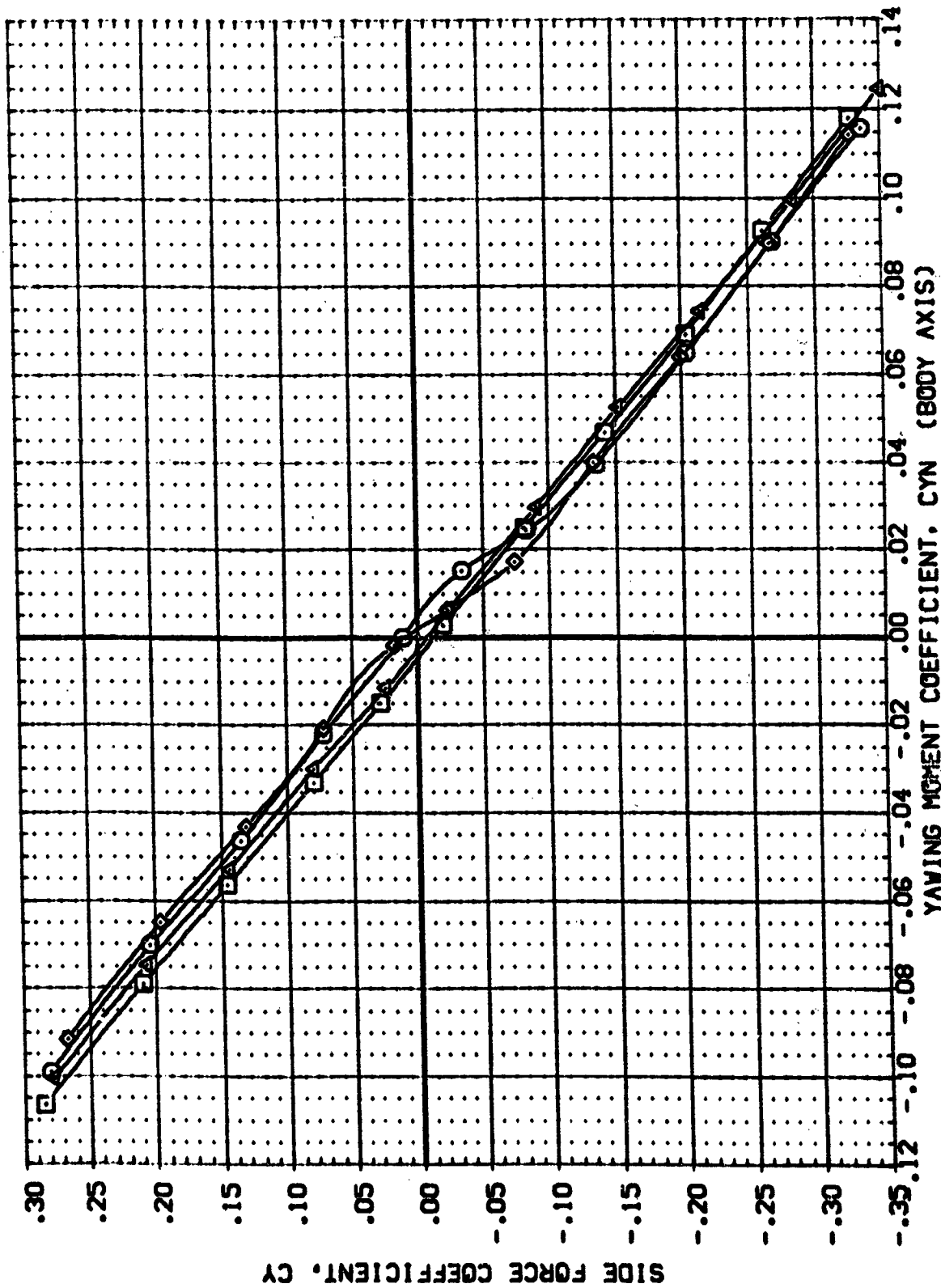


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.



DATA SET SYMBOL: (RB7006) (RB7007) (RB7008) (RB7009)  
 CONFIGURATION DESCRIPTION: AVES 3.5-169 IA10 CS T10 AT2 PLUVE ON (RB7006) AVES 3.5-169 IA10 CS T10 AT2 PLUVE OFF (RB7007) AVES 3.5-169 IA10 CS T10 AT2 PLUVE ON (RB7008) AVES 3.5-169 IA10 CS T10 AT2 PLUVE OFF (RB7009)

ALPHA: .000 .000 .000 .000  
 AILRON: .000 .000 .000 .000  
 ELEVON: .000 .000 .000 .000  
 RUDDER: 10.000 10.000 .000 .000  
 REFERENCE INFORMATION: SREF 2690.0000 SQ.FT. LREF 1290.0000 IN. BREF 936.6800 IN. XMRP 1076.4800 IN. YMRP 400.0000 IN. ZMRP .0100 IN. SCALE

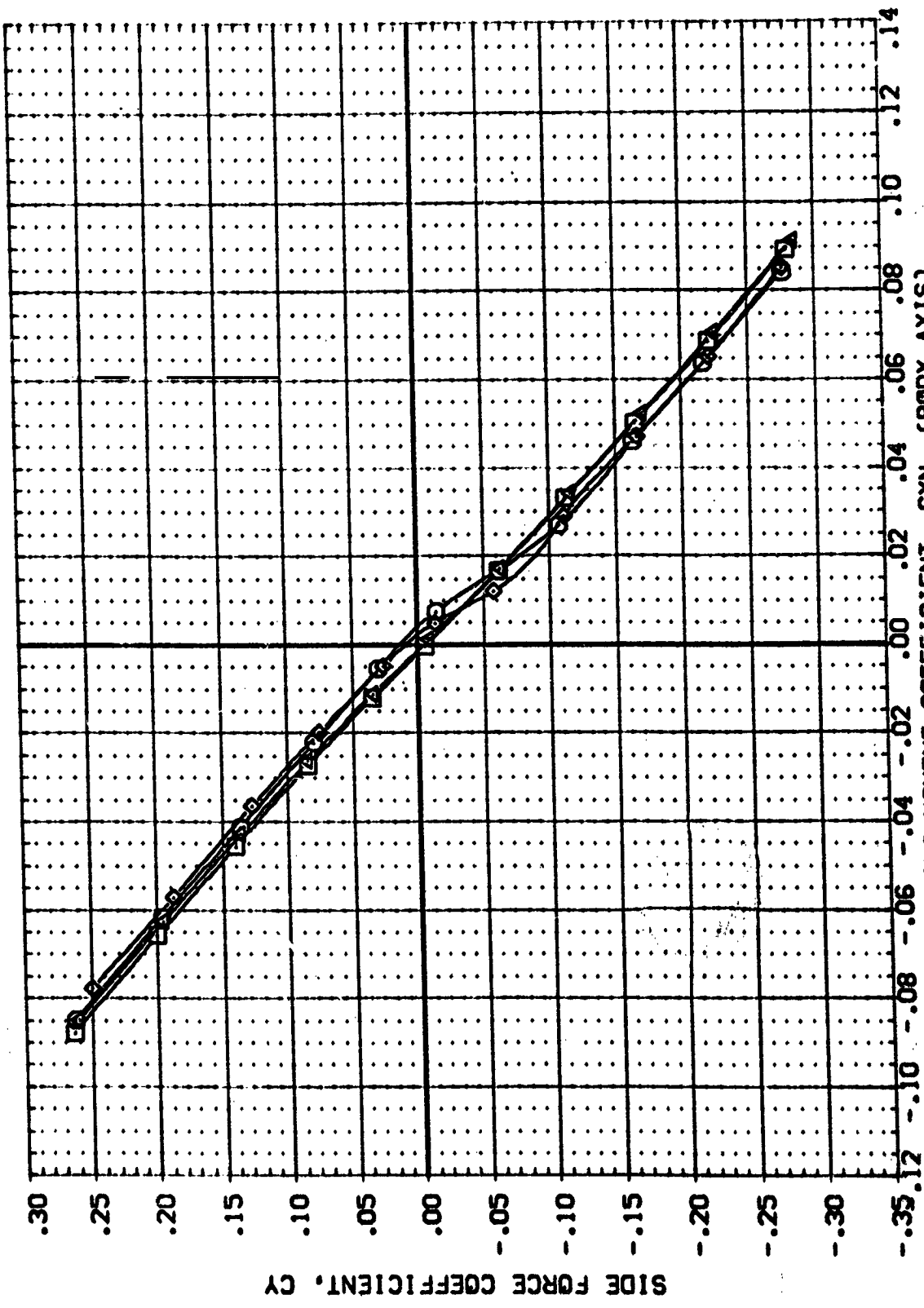


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(B)MACH = 7.32

DATA SET SYMBL. CONFIGURATION DESCRIPTION  
 (087011) AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF  
 (157005) AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF  
 (157001) DATA NOT AVAILABLE  
 (157005) DATA NOT AVAILABLE

BETA      AILTRON      ELEVON      ROLLER  
 .000      .000      .000      .000  
 .000      10.000      .000      .000  
 .000      10.000      .000      .000

REFERENCE INFORMATION      SQ. FT.  
 SREF      2690.0000      IN.  
 LREF      1290.0000      IN.  
 BREF      936.6800      IN.  
 XMRP      1076.4800      IN.  
 YMRP      400.0000      IN.  
 ZMRP      400.0000      IN.  
 SCALE      .0100

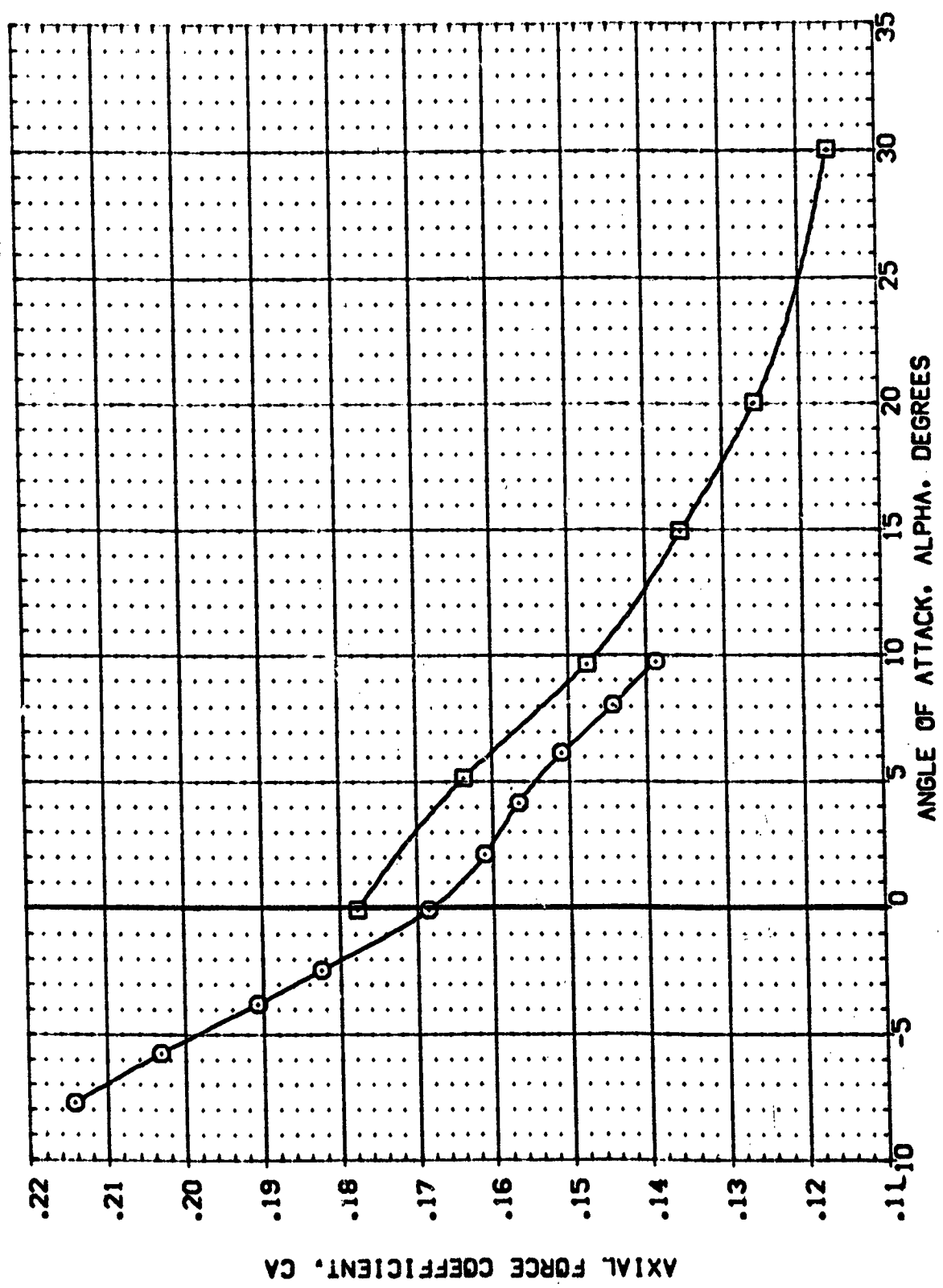


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(M)MACH = 5.26

DATA SET SYMB. CONFIGURATION DESCRIPTION  
 (DB7011) DATA NOT AVAILABLE  
 (SB7005) DATA NOT AVAILABLE  
 (RB7001) AMES 3.5-169 1A10 OS T10 AT2 PLUVE OFF  
 (TB7005) AMES 3.5-169 1A10 OS T10 AT2 PLUVE OFF

BETA .000  
 AILRON 10.000  
 ELEVON .000  
 RUDDER .000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1790.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP 400.0000 IN.  
 ZMRP .0100 IN.  
 SCALE

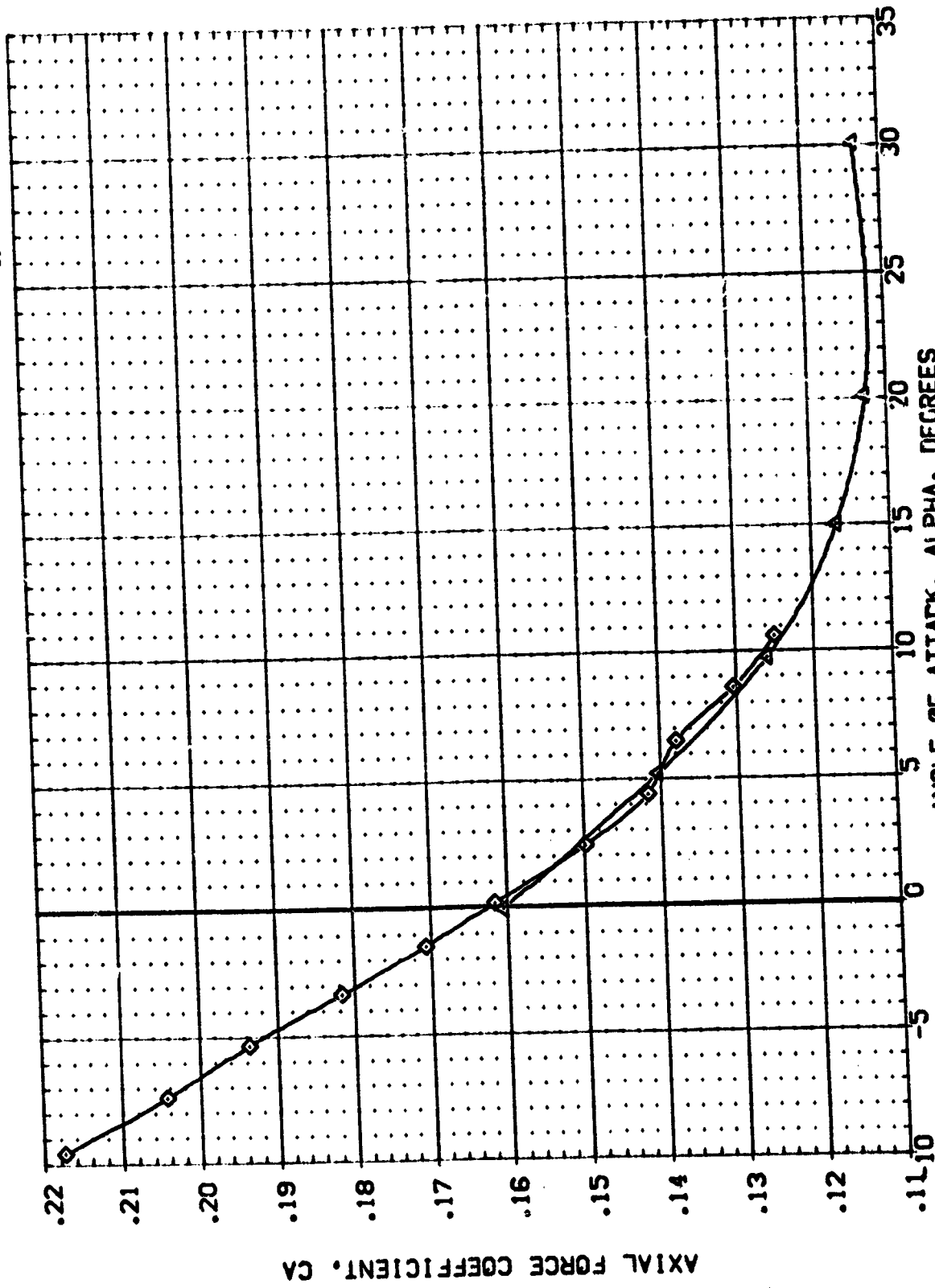


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(B)MACH = 7.32

DATA SET SYMBOL: (087011) (987005) (887001) (187005)  
 CONFIGURATION DESCRIPTION: AVES 3-5-169 IA10 09 T10 AT2 PLUME OFF; AVES 3-5-169 IA10 09 T10 AT2 PLUME OFF; DATA NOT AVAILABLE; DATA NOT AVAILABLE  
 BETA: 0.00; .00; .00; .00  
 AIRLON: .000; 10.000; .000; 10.000  
 ELEVON: .000; .000; .000; .000  
 RUDDER: .000; .000; .000; .000  
 REFERENCE INFORMATION: SREF 2680.0000 SO.FT. 50.0000; LREF 1250.0000 IN. 12.5000; BREF 936.6800 IN. 9.3668; XMRP 1076.4800 IN. 10.7648; YMRP 400.0000 IN. 4.0000; SCALE .0100

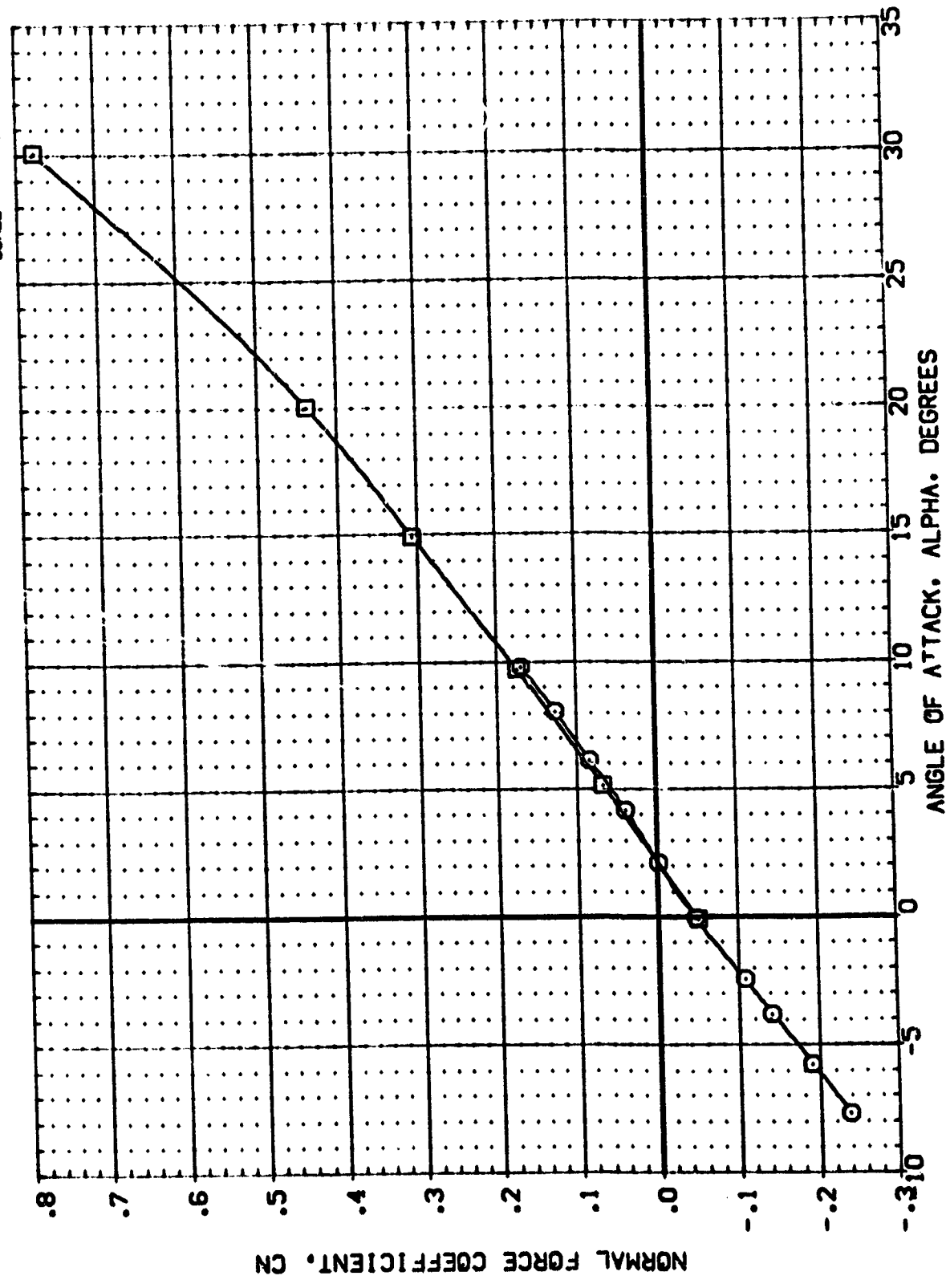


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(A)MACH = 5.26

DATA SET SYMBOL: (DB7011), (SB7005), (RB7001), (TB7005)

CONFIGURATION DESCRIPTION: DATA NOT AVAILABLE, DATA NOT AVAILABLE, ARES 3-5-169 IA10, ARES 3-5-169 IA10

OS T10 AT2 PLUME OFF, OS T10 AT2 PLUME OFF

BETA: .000, .000, .000, .000

AILERON: .000, 10.000, .000, 10.000

ELEVON: .000, .000, .000, .000

RUDDER: .000, .000, .000, .000

REFERENCE INFORMATION: SREF 2690.0000 SQ.FT., LREF 1290.0000 IN., BREF 936.6800 IN., YMRP 1076.4800 IN., ZMRP 400.0000 IN., SCALE .0100

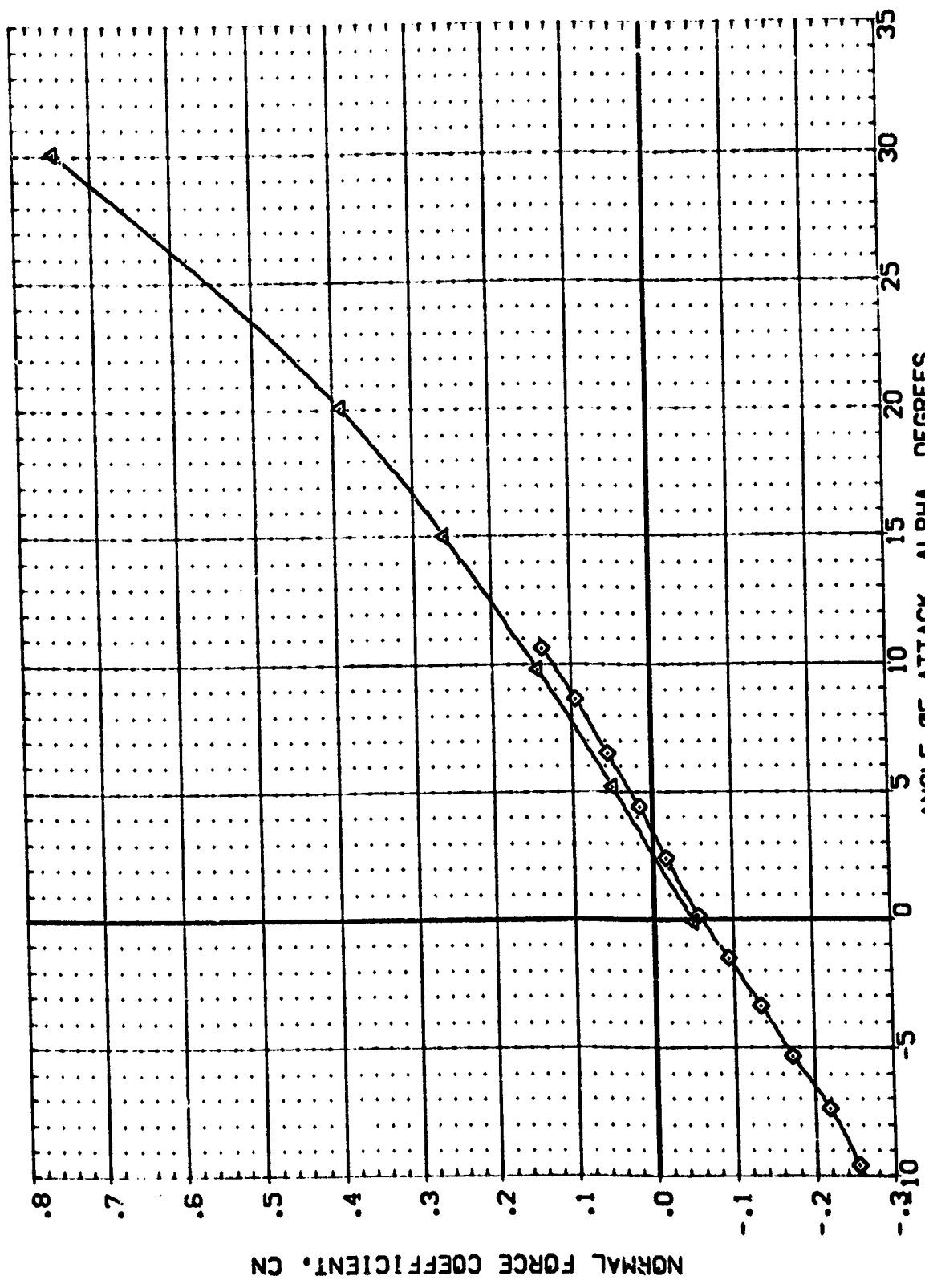


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(B)MACH = 7.32



(08701) DATA NOT AVAILABLE 09 110 AT2 PLUME OFF  
 (S8700) DATA NOT AVAILABLE 09 110 AT2 PLUME OFF  
 (R5700) AVES 3.5-169 1A10 09 110 AT2 PLUME OFF  
 (T8700) AVES 3.5-169 1A10 09 110 AT2 PLUME OFF

SREF .000  
 LREF .000  
 XMRP .000  
 YMRP .000  
 ZMRP .000  
 SCALE .0100

2680.0000 SQ.FT.  
 1250.0000 IN.  
 936.6800 IN.  
 1076.4800 IN.  
 400.0000 IN.  
 400.0000 IN.

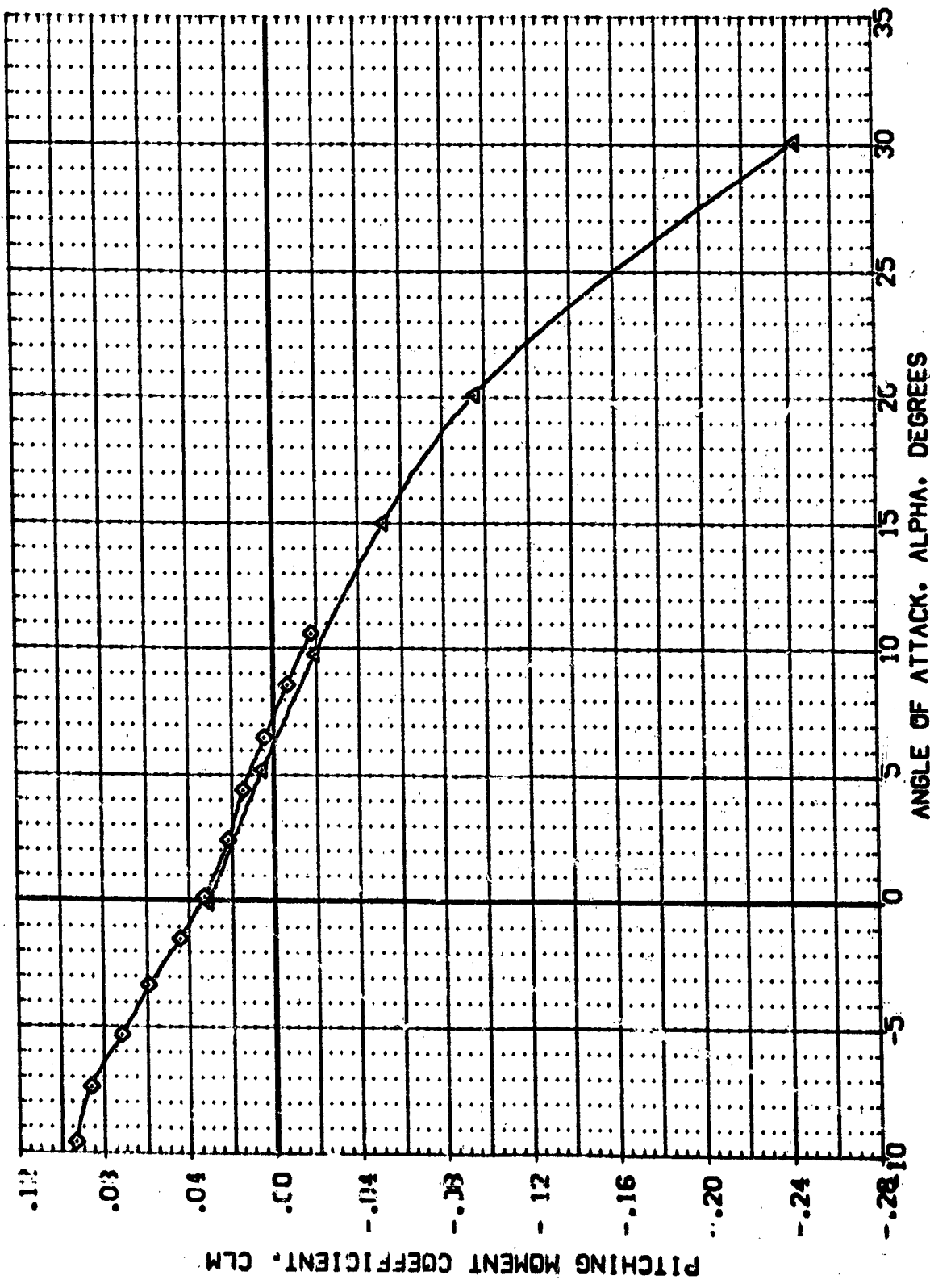


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(8)MACH = 7.32

DATA SET SYMBL. CONFIGURATION DESCRIPTION  
 (087011) ARES 3.5-169 IA10 08 T10 AT2 PLUVE OFF  
 (587005) ARES 3.5-169 IA10 08 T10 AT2 PLUVE OFF  
 (187001) DATA NOT AVAILABLE  
 (187005) DATA NOT AVAILABLE

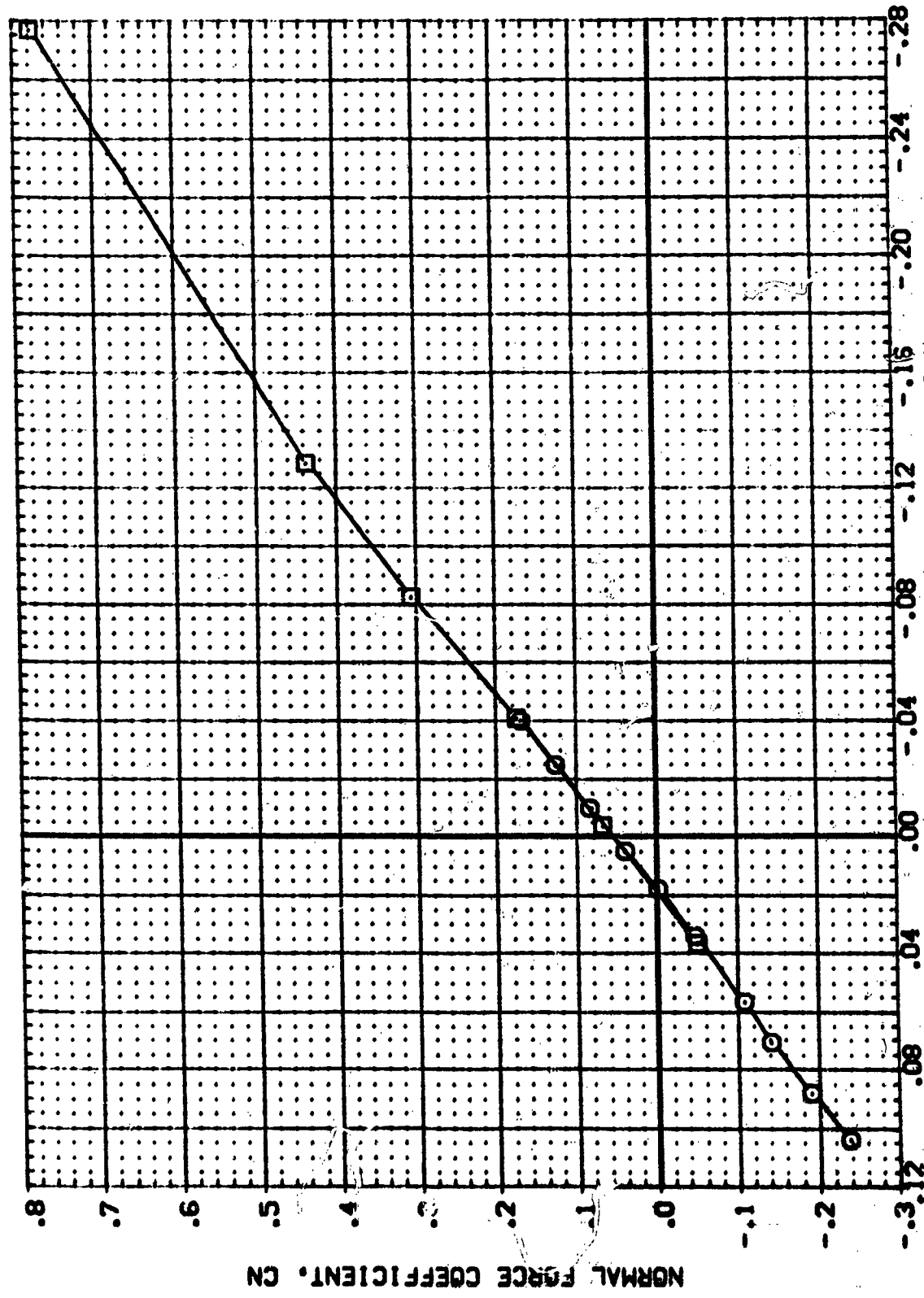
BETA .000 .000 .000  
 .000 10.000 .000  
 .000 10.000 .000

AILRON .000 .000 .000  
 10.000 .000 .000  
 10.000 .000 .000

ELEVEN .000 .000 .000  
 .000 .000 .000  
 .000 .000 .000

RUDDER .000 .000 .000  
 .000 .000 .000  
 .000 .000 .000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP 400.0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100



PITCHING MOMENT COEFFICIENT, CLM

NORMAL FORCE COEFFICIENT, CN

FIG. 7 EFFECT OF DIFFERENTIAL ELEVEN DEFLECTION.

(A)MACH = 5.26



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (187005) □ DATA NOT AVAILABLE  
 (187005) □ DATA NOT AVAILABLE  
 (187005) □ AVES 3.5-169 IA1D OS T10 ATZ PLUVE OFF  
 (187005) □ AVES 3.5-169 IA1D OS T10 ATZ PLUVE OFF

BETA .000  
 AILRON .000  
 ELEVON .000  
 RUDDER .000  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 YPRP 1076.4800 IN.  
 ZPRP 400.0000 IN.  
 SCALE .0100

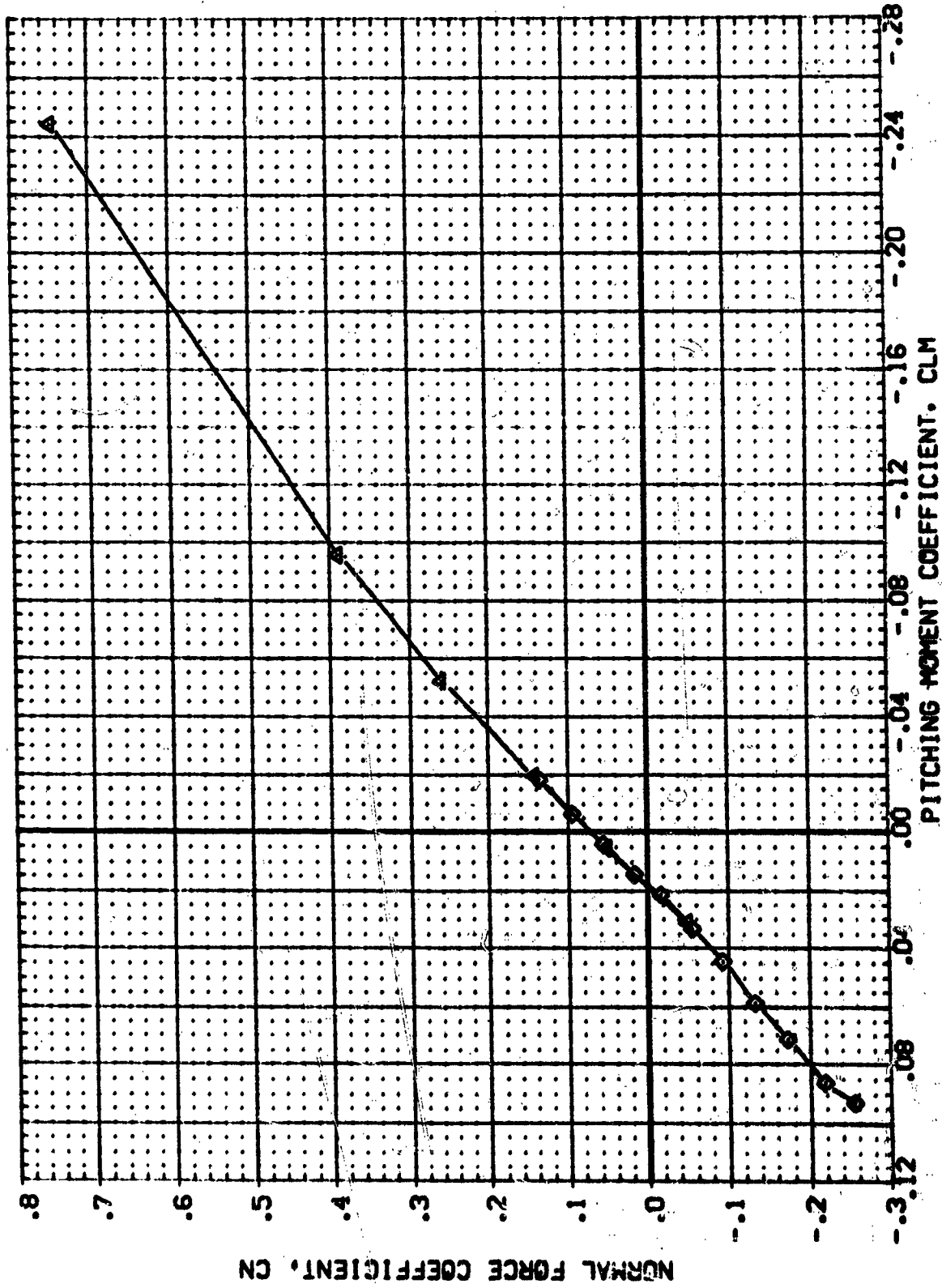


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(B)MACH = 7.32

DATA SET SYMBOL: (D87011) (S87005) (R87001) (T87005)

CONFIGURATION DESCRIPTION: APES 3-5-169 IA10 OS T19 AT2 PLUME OFF APES 3-5-169 IA10 OS T10 AT2 PLUME OFF DATA NOT AVAILABLE DATA NOT AVAILABLE

BETA: .000 .000 .000 .000  
AILRON: .000 10.000 .000 10.000  
ELEVON: .000 .000 .000 .000  
RUDDER: .000 .000 .000 .000

REFERENCE INFORMATION: SREF 2690.0000 SQ.FT. LREF 1290.0000 IN. BREF 936.6800 IN. XMRP 1076.4800 IN. YMRP 400.0000 IN. ZMRP .0100 IN. SCALE

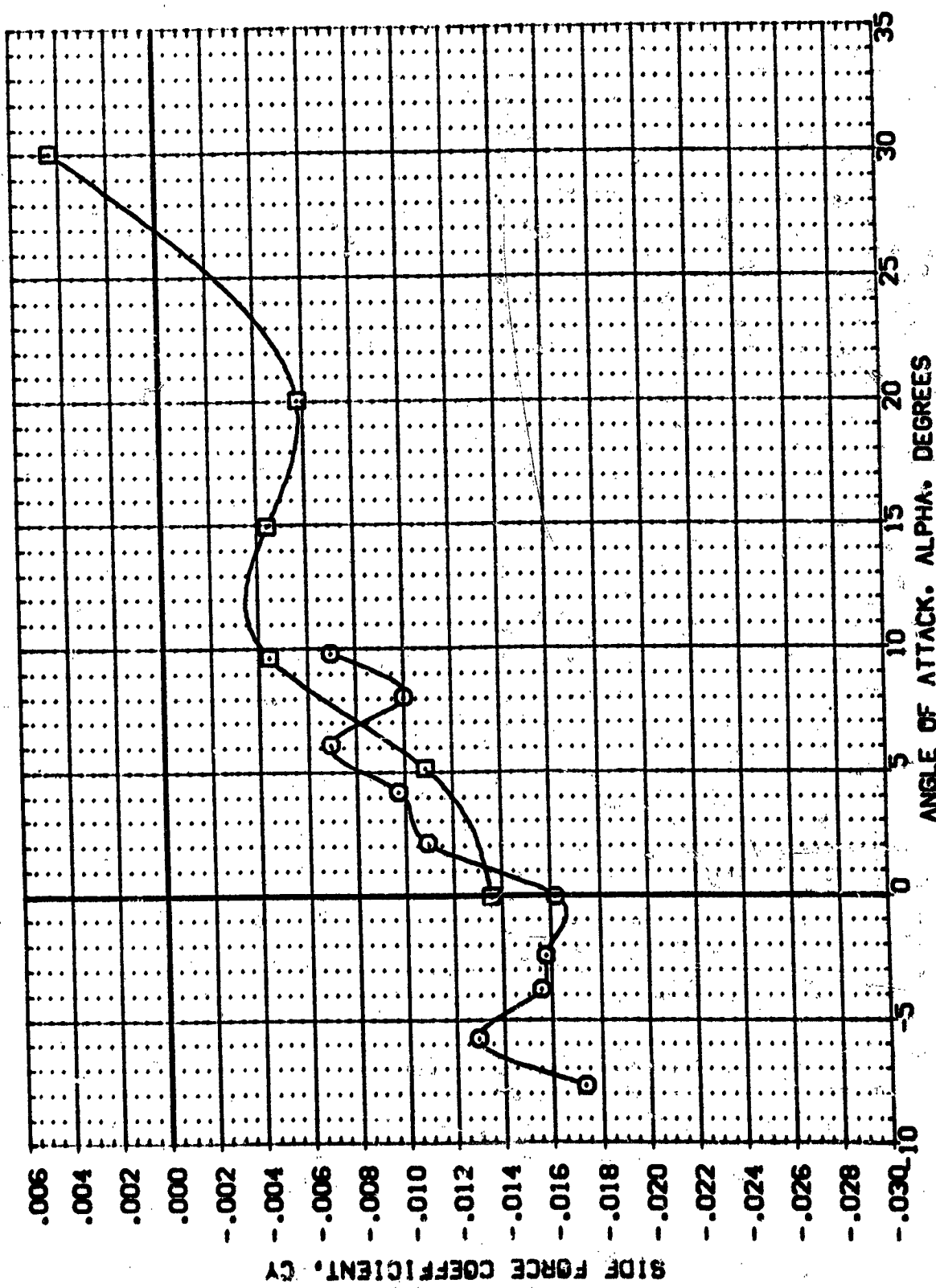


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(A)MACH = 5.26

DATA SET SYMBL CONFIGURATION DESCRIPTION  
 (087011) DATA NOT AVAILABLE  
 (S87005) DATA NOT AVAILABLE  
 (R87001) 09 T10 AT2 PLUVE OFF  
 (T87005) 09 T10 AT2 PLUVE OFF  
 AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1250.0000 IN.  
 XMRP 936.6800 IN.  
 YMRP 1076.4800 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

BETA .000  
 AILRON .000  
 ELEVON .000  
 RUDDER .000

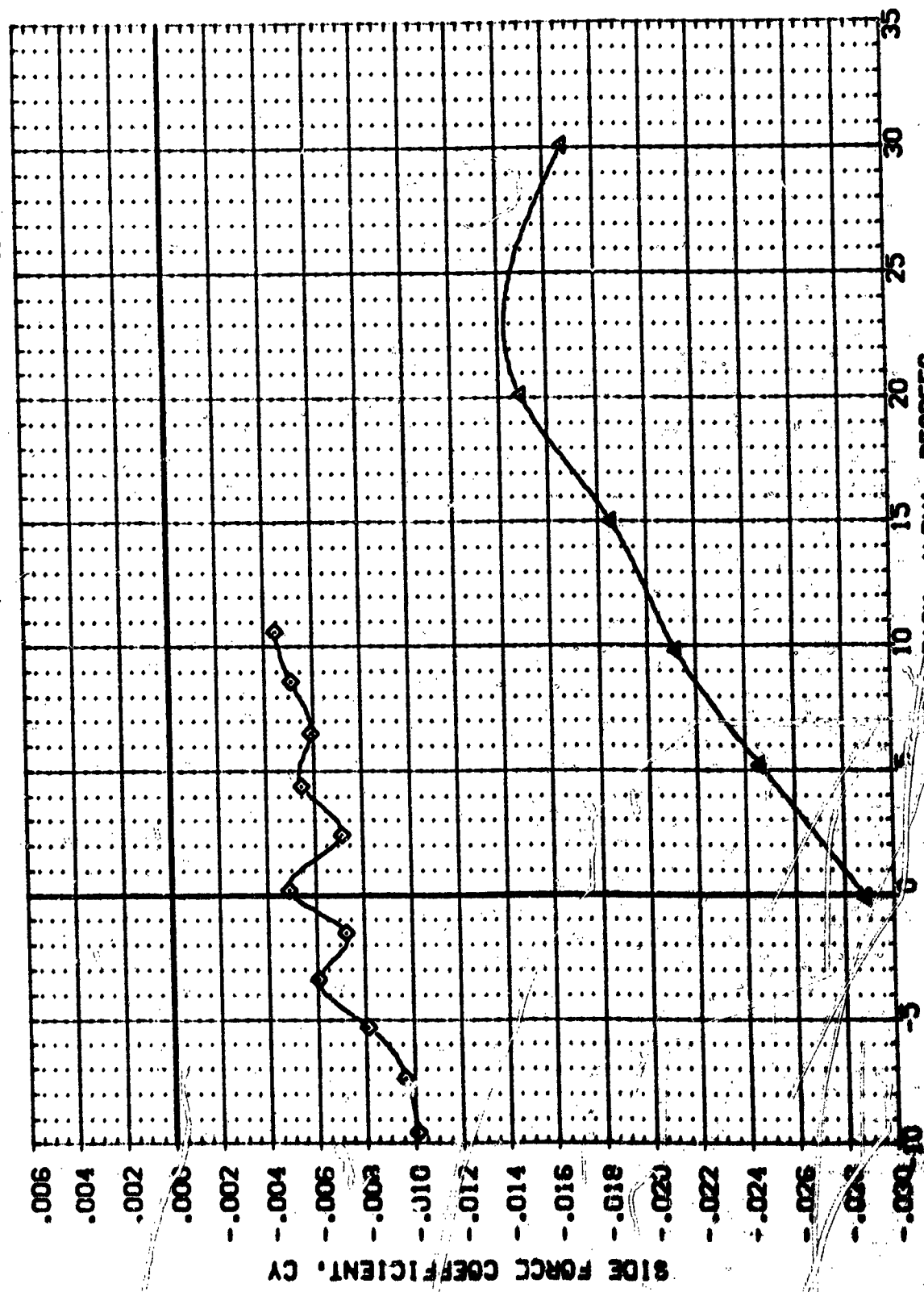


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

ATTACH - 7.32

BETA .000  
 AILSON .000  
 ELEVON .000  
 ROLLER .000  
 REFERENCE INFORMATION  
 SREF 2680.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BRFP 936.6800 IN.  
 YMRP 1076.4800 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

DATA SET SYMBOL: (087011), (S87005), (R87001), (T87005)  
 CONFIGURATION DESCRIPTION: 08 T10 AT2 PLUPE OFF, 09 T10 AT2 PLUPE OFF  
 DATA NOT AVAILABLE

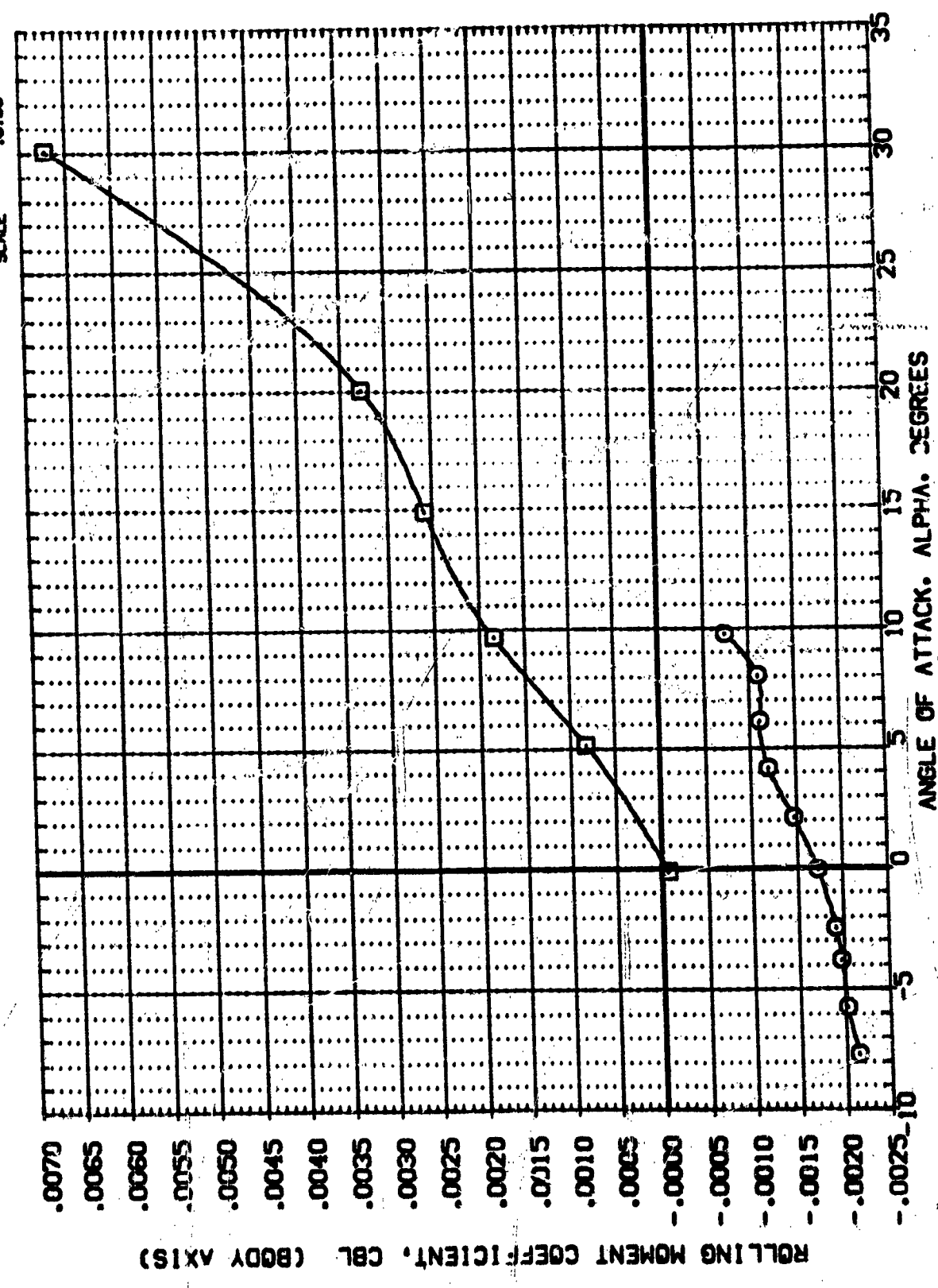


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

DATA SET SYMBOL: (DB7011)  
 (SB7005)  
 (RB7001)  
 (TB7005)

CONFIGURATION DESCRIPTION:  
 DATA NOT AVAILABLE  
 AXES 3-5-169 IAI0 CS T10 AT2 PLUME OFF  
 AXES 3-5-169 IAI0 CS T10 AT2 PLUME OFF

BETA: .000  
 .000  
 .000

AILERON: .000  
 10.000  
 10.000

ELEVON: .000  
 .000  
 .000

RUDER: .000  
 .000  
 .000

REFERENCE INFORMATION:  
 SREF: 2650.0000 SQ.FT.  
 LRET: 1250.0000 IN.  
 BREF: 936.6800 IN.  
 YMRP: 1076.4800 IN.  
 ZMRP: 400.0000 IN.  
 SCALE: .0100

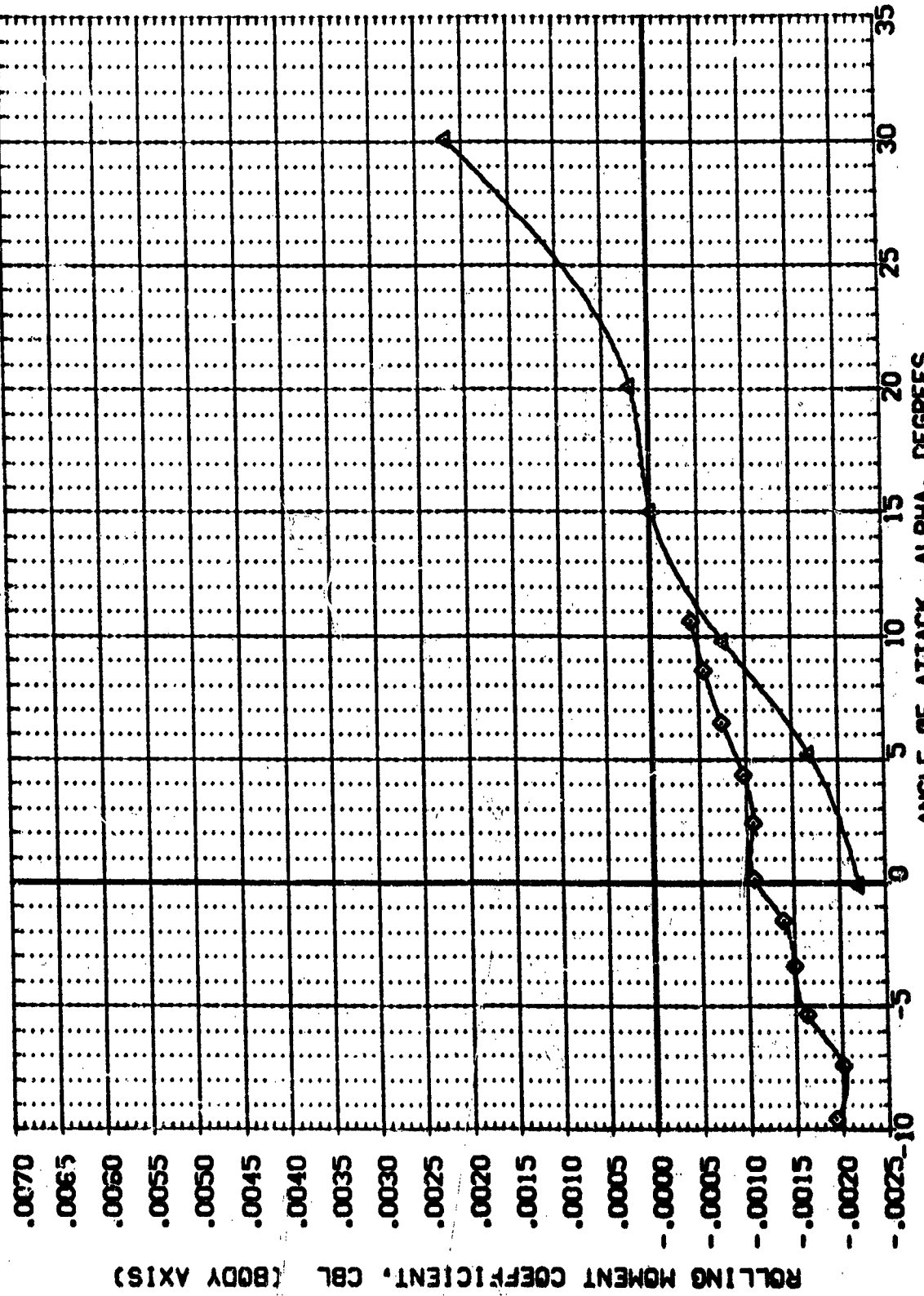


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(B)MACH = 7.32

DATA SET SYMBO  
 (087011)  
 (587005)  
 (587001)  
 (787005)

CONFIGURATION DESCRIPTION  
 AVES 3.5-169 IA10 OS T10 AT2 PLUVE OFF  
 AVES 3.5-169 IA10 OS T10 AT2 PLUVE OFF  
 DATA NOT AVAILABLE  
 DATA NOT AVAILABLE

BETA  
 .000  
 .000  
 .000  
 .000

AILERON  
 .000  
 10.000  
 .000  
 10.000

ELEVON  
 .000  
 .000  
 .000  
 .000

RUDDER  
 .000  
 .000  
 .000  
 .000

REFERENCE INFORMATION  
 SREF 2680.0000 SO.FT.  
 LREF 1250.0000 IN.  
 BRFP 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP 400.0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

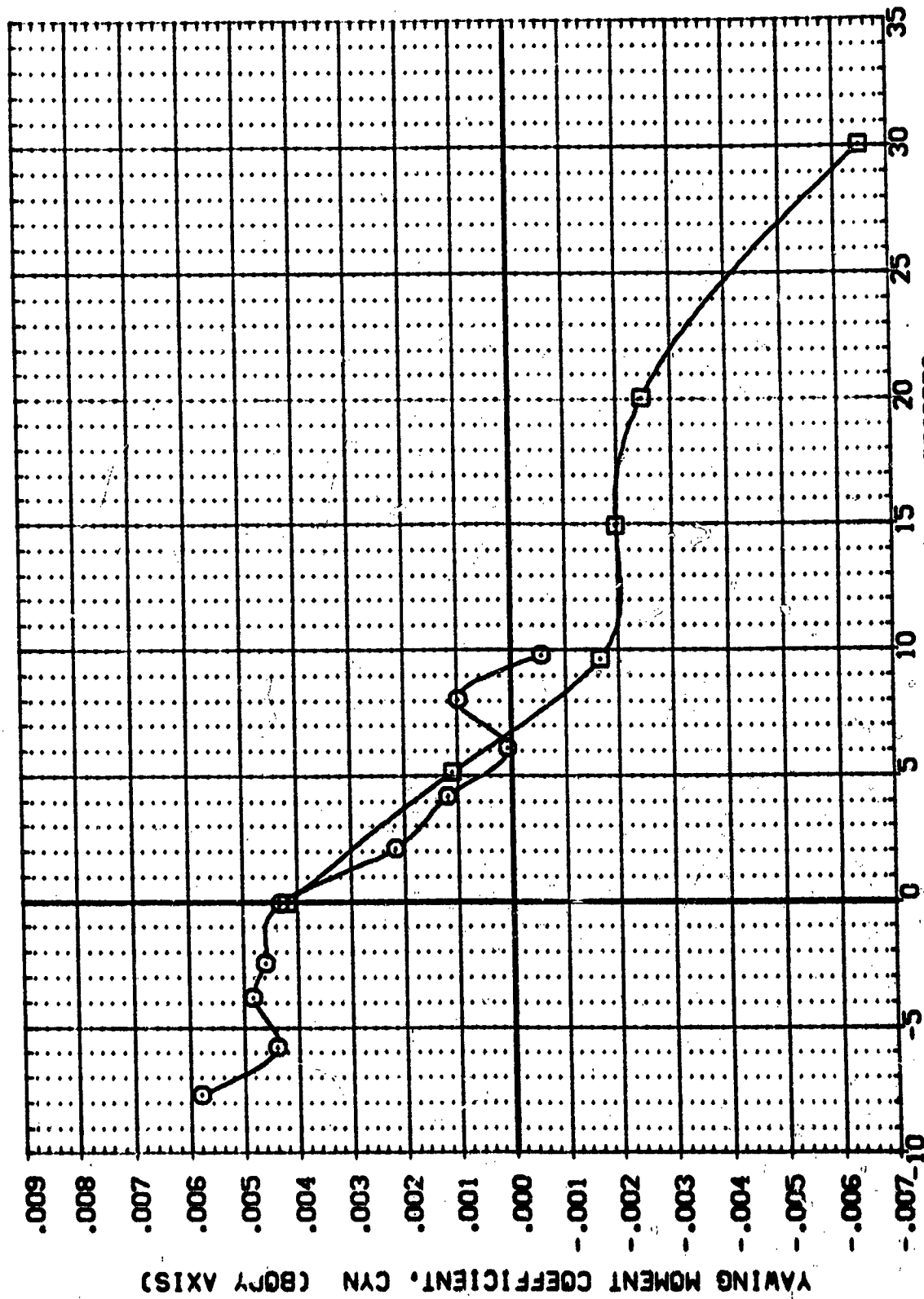


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(A)MACH = 5.26

DATA SET SYMBOL: (DB7011) (SB7005) (RB7001) (TB7005)  
 CONFIGURATION DESCRIPTION: DATA NOT AVAILABLE  
 OS T10 AT2 PLUVE OFF  
 OS T10 AT2 PLUVE OFF  
 AVES 3.5-189 1A10  
 AVES 3.5-189 1A10

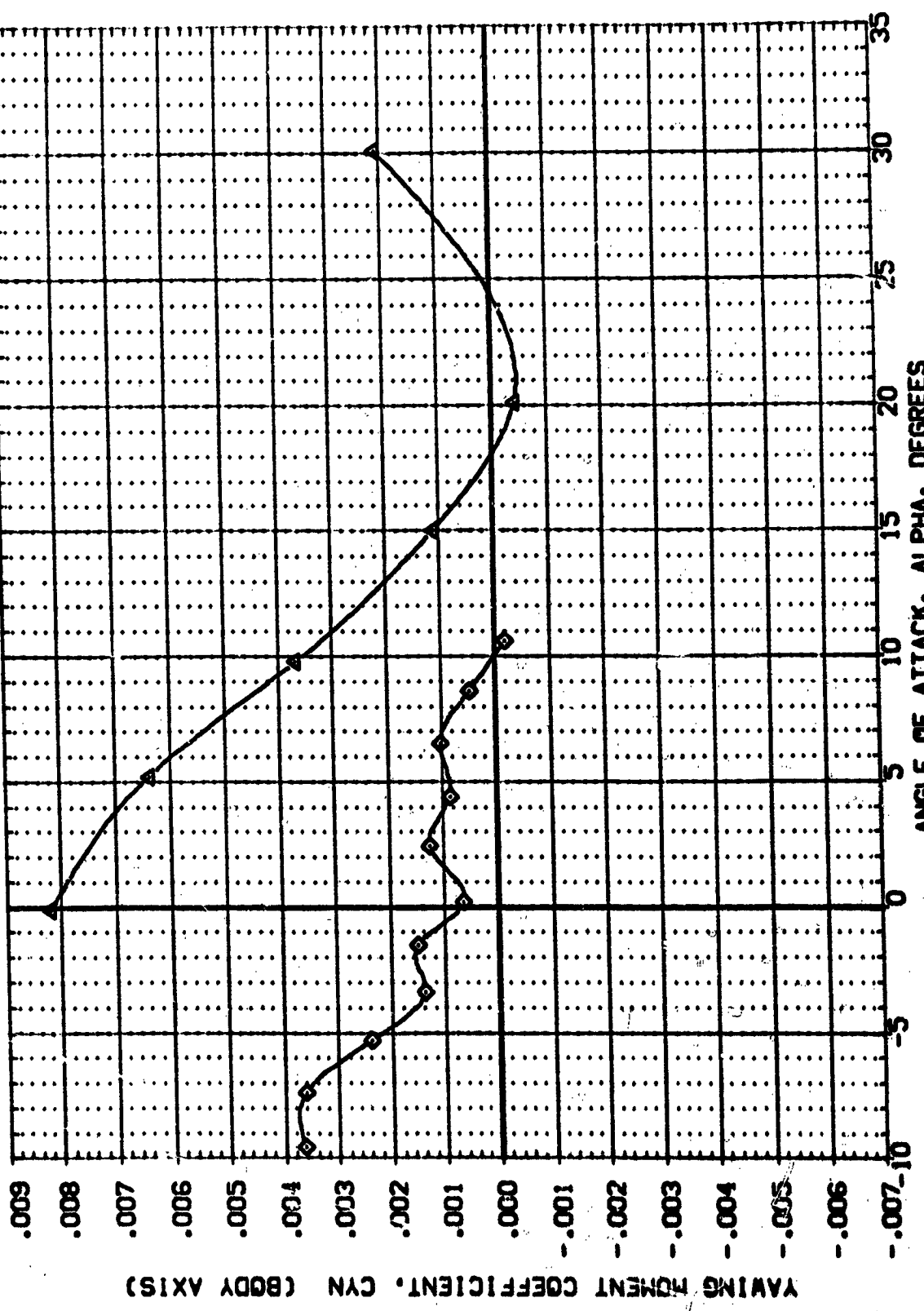


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(B)MACH = 7.32

DATA SET SYMBOL (REF7006) (REF7007)

CONFIGURATION DESCRIPTION  
 AVES 3.5-169 IA10 OS T10 AT2 PLANE OFF  
 AVES 3.5-169 IA10 OS T10 A:2 PLANE OFF

ALPHA .000  
 AIRLON .000  
 ELEVON .000  
 RUDDER .000  
 SREF 2690.0000  
 LREF 1250.0000  
 BREF 956.6800  
 XMRP 1076.4800  
 YMRP 400.0000  
 ZMRP 400.0000  
 SCALE .0100

REFERENCE INFORMATION  
 SO.FT. IN.  
 IN.  
 IN.  
 IN.  
 IN.

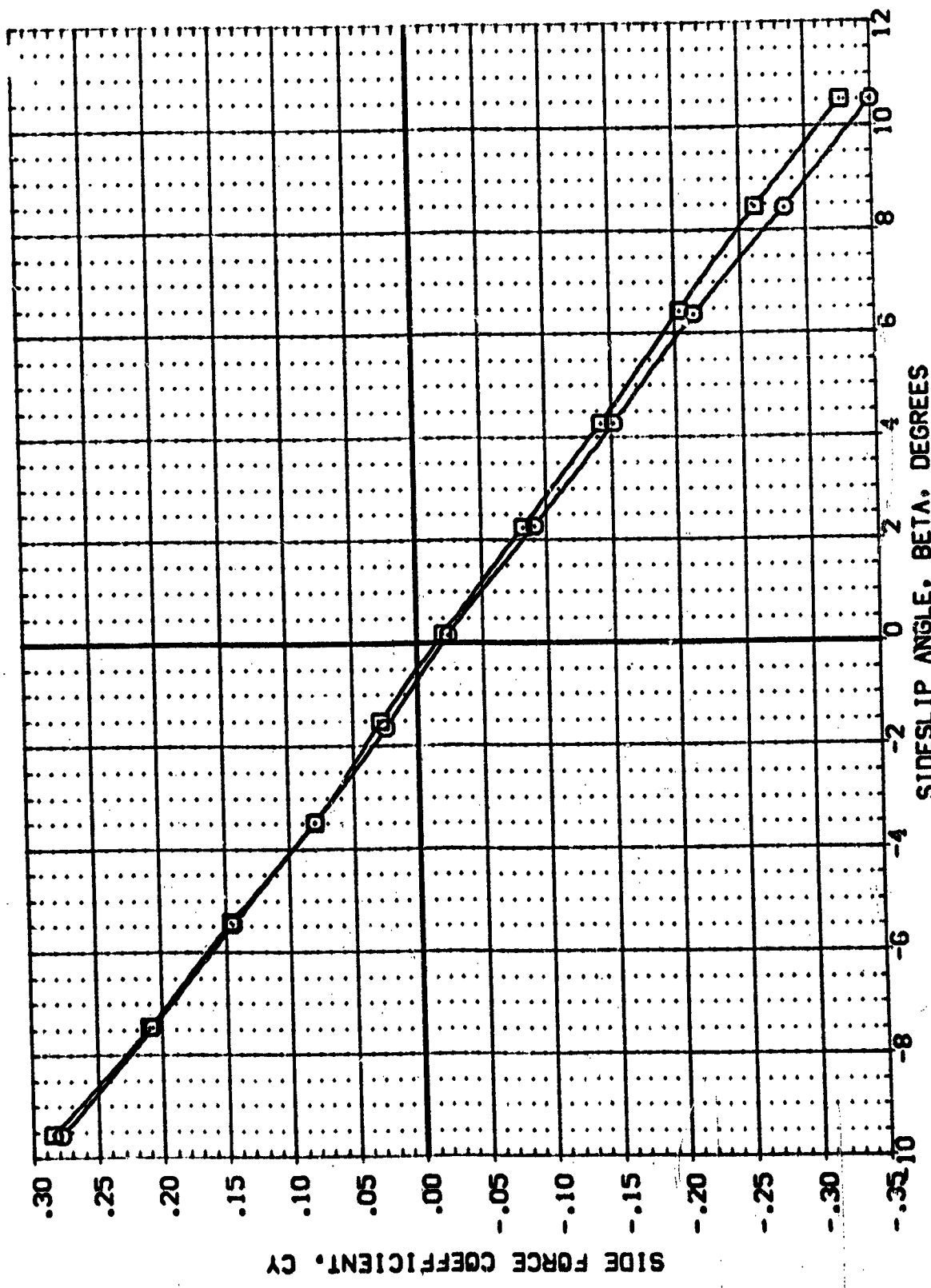


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26



DATA SET SYMBL	CONFIGURATION DESCRIPTION	ALPHA	A1LRN	ELEVON	RUDDER	REFERENCE INFORMATION
(R97005)	AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF	.000	.000	.000	.000	SREF 2690.0000 SO.FT.
(R97007)	AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
						BREF 936.6800 IN.
						XPRP 1076.4800 IN.
						YPRP .0000 IN.
						Z-PRP .0000 IN.
						SCALE .0100

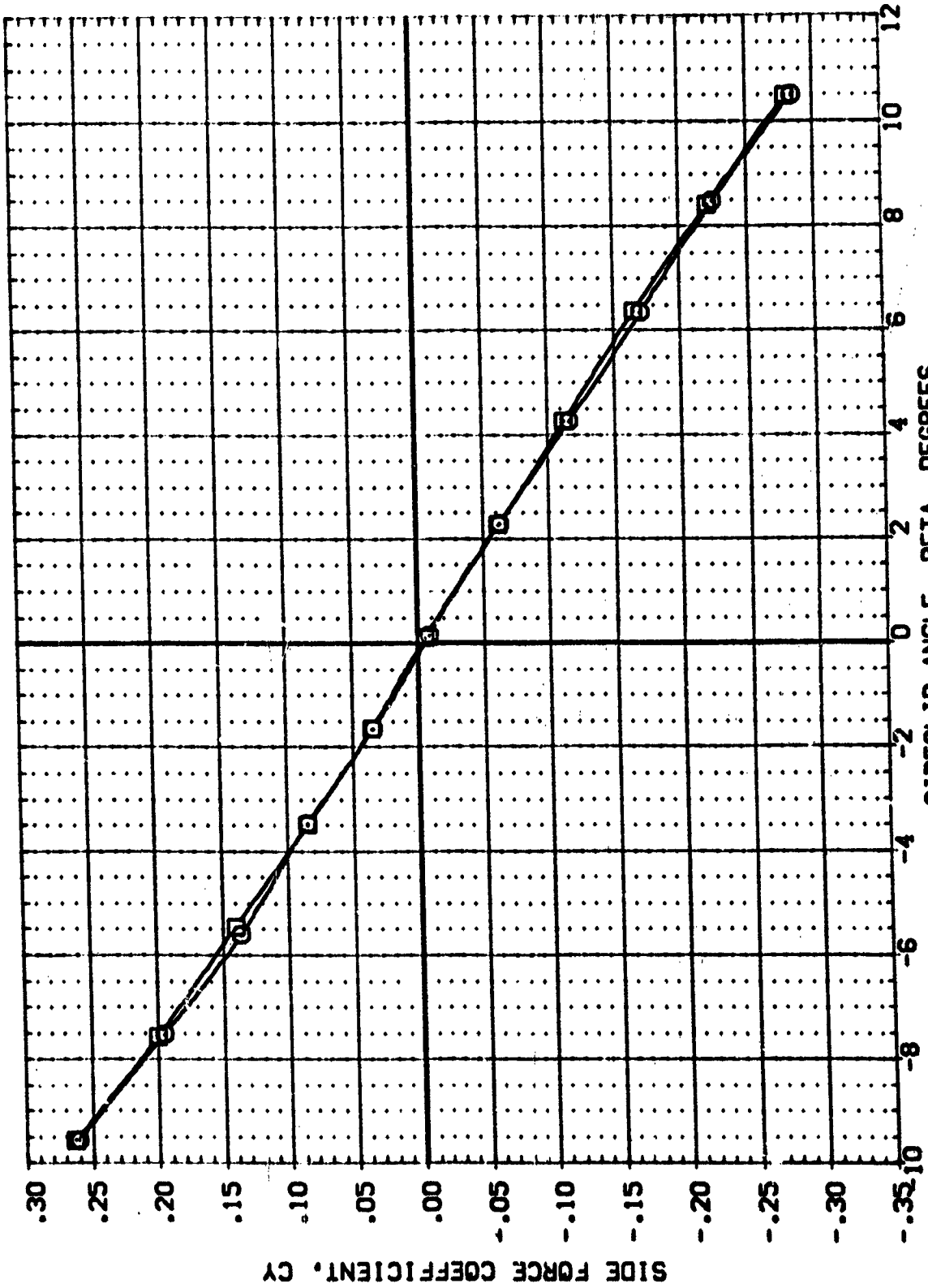


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBOL (RB7006) (RB7007) □

CONFIGURATION DESCRIPTION  
 APES 3.5-169 IA10 09 T10 AT2 PLUVE OFF  
 APES 3.5-169 IA10 09 T10 AT2 PLUVE OFF

ALPHA AILRON ELEVON RUDDER  
 .000 .000 .000 .000  
 .000 .000 .000 .000

REFERENCE INFORMATION  
 SREF 1690.0000 SO.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XPRP 1376.4800 IN.  
 YPRP .0000 IN.  
 ZPRP 400.0000 IN.  
 SCALE 400.0100

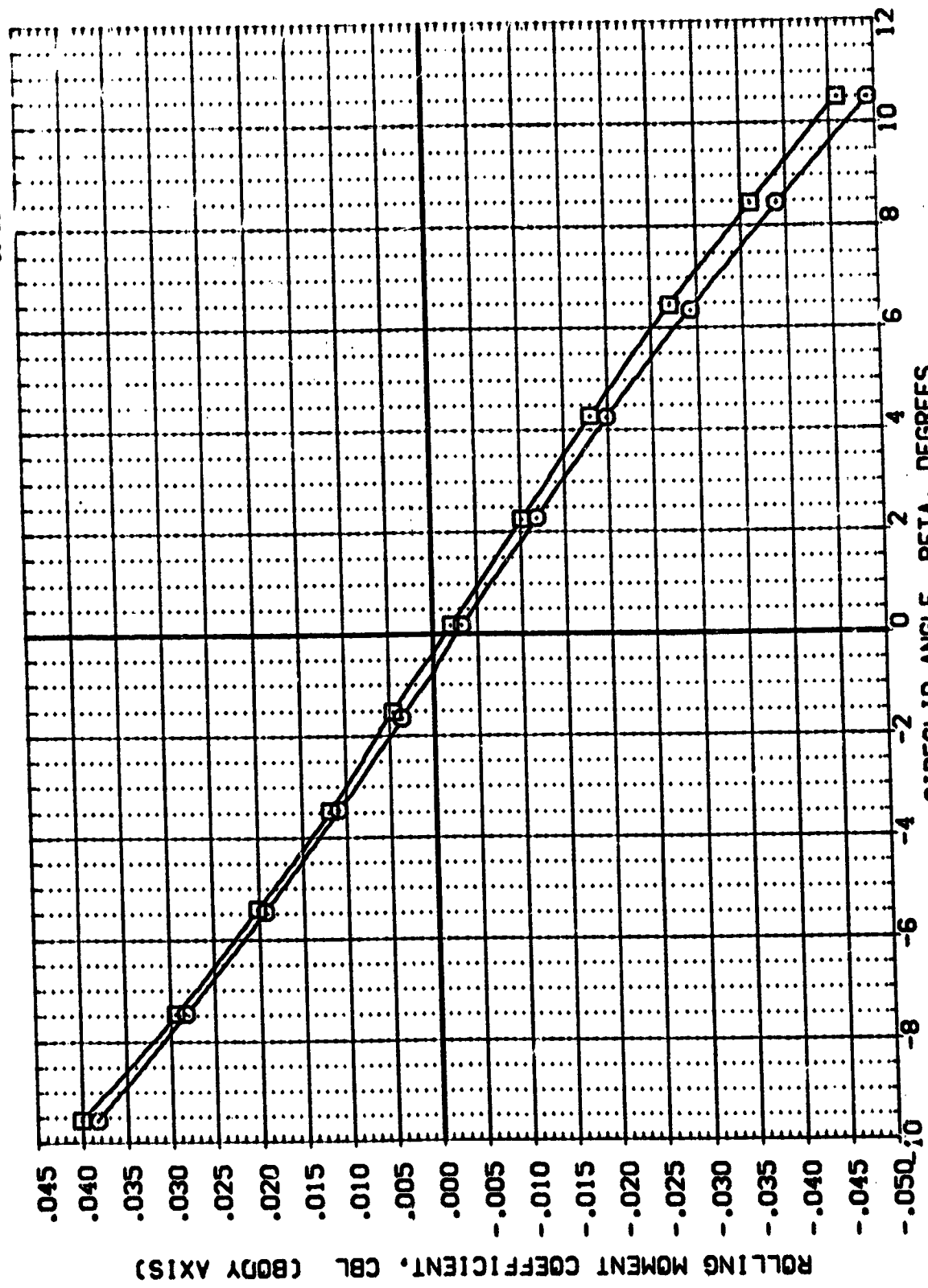


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBOL: (RB7006) (RB7007)

CONFIGURATION DESCRIPTION: AMES 3.5-169 IA10 OS T10 AT2 PLUME OFF

ALPHA: .000 .000

AILERON: .000 .000

ELEVON: .000 .000

RUDDER: .000 10.000

REFERENCE INFORMATION: SREF 2690.0000 SQ.FT. LREF 1790.0000 IN. BREF 936.6800 IN. XMRP 1076.4800 IN. YMRP .0000 IN. ZMRP 400.0000 IN. SCALE .0100

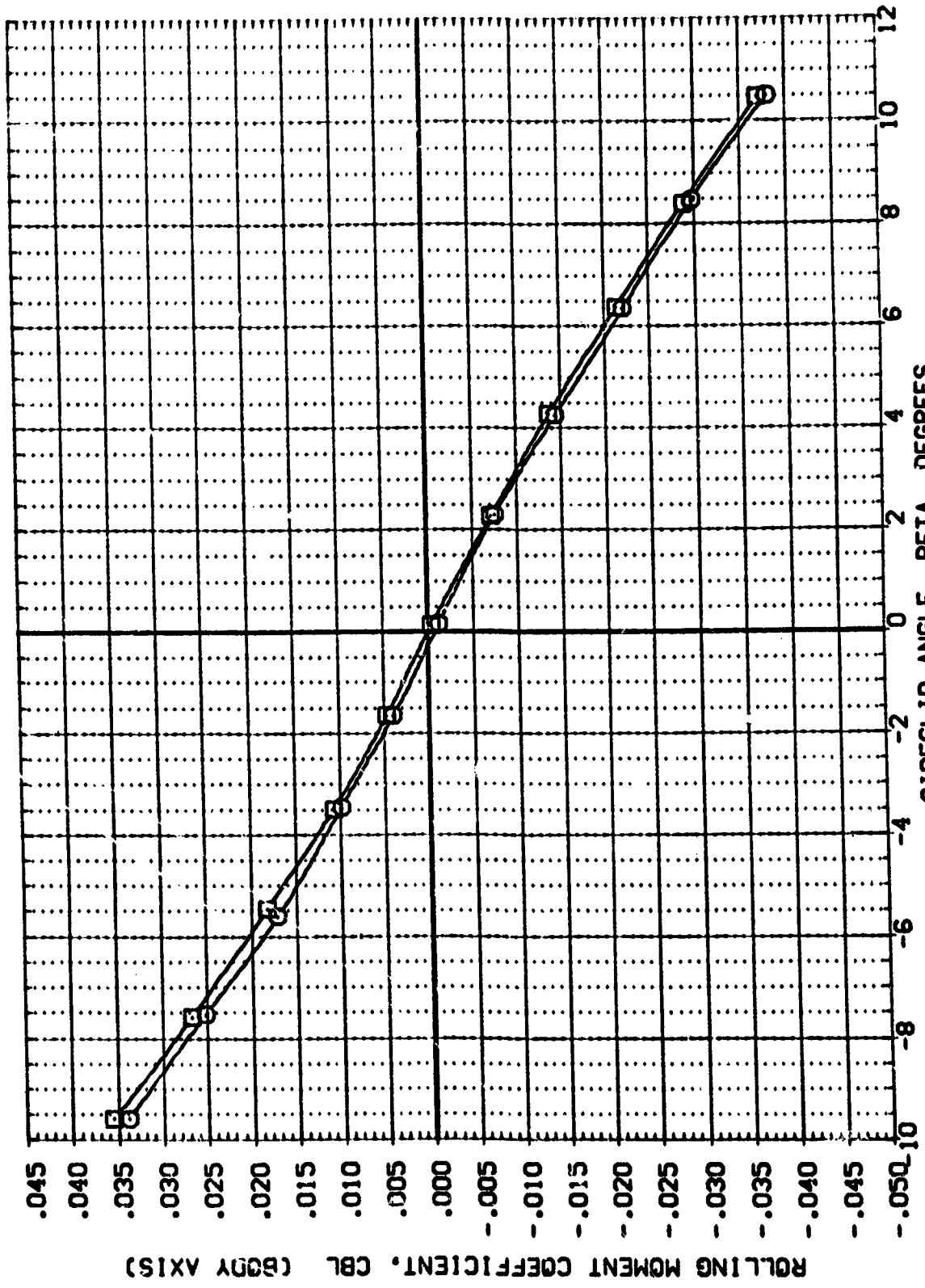


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B) MACH = 7.32

DATA SET SYMBOL (R87006) (R87007)

CONFIGURATION DESCRIPTION  
 AVES 3.5-189 IA10 OS T10 AT2 PLUVE OFF  
 AVES 3.5-189 IA10 OS T10 AT2 PLUVE OFF

ALPHA .000  
 AIRLON .000  
 ELEVON .000  
 RUDDER .000  
 SCALE .0100

REFERENCE INFORMATION  
 SREF 2690.0000 SD.FT.  
 LREF 1290.0000 IN.  
 XMRP 936.6800 IN.  
 YMRP 1076.4800 IN.  
 ZMRP .0000 IN.  
 SCALE 400.0000 IN.

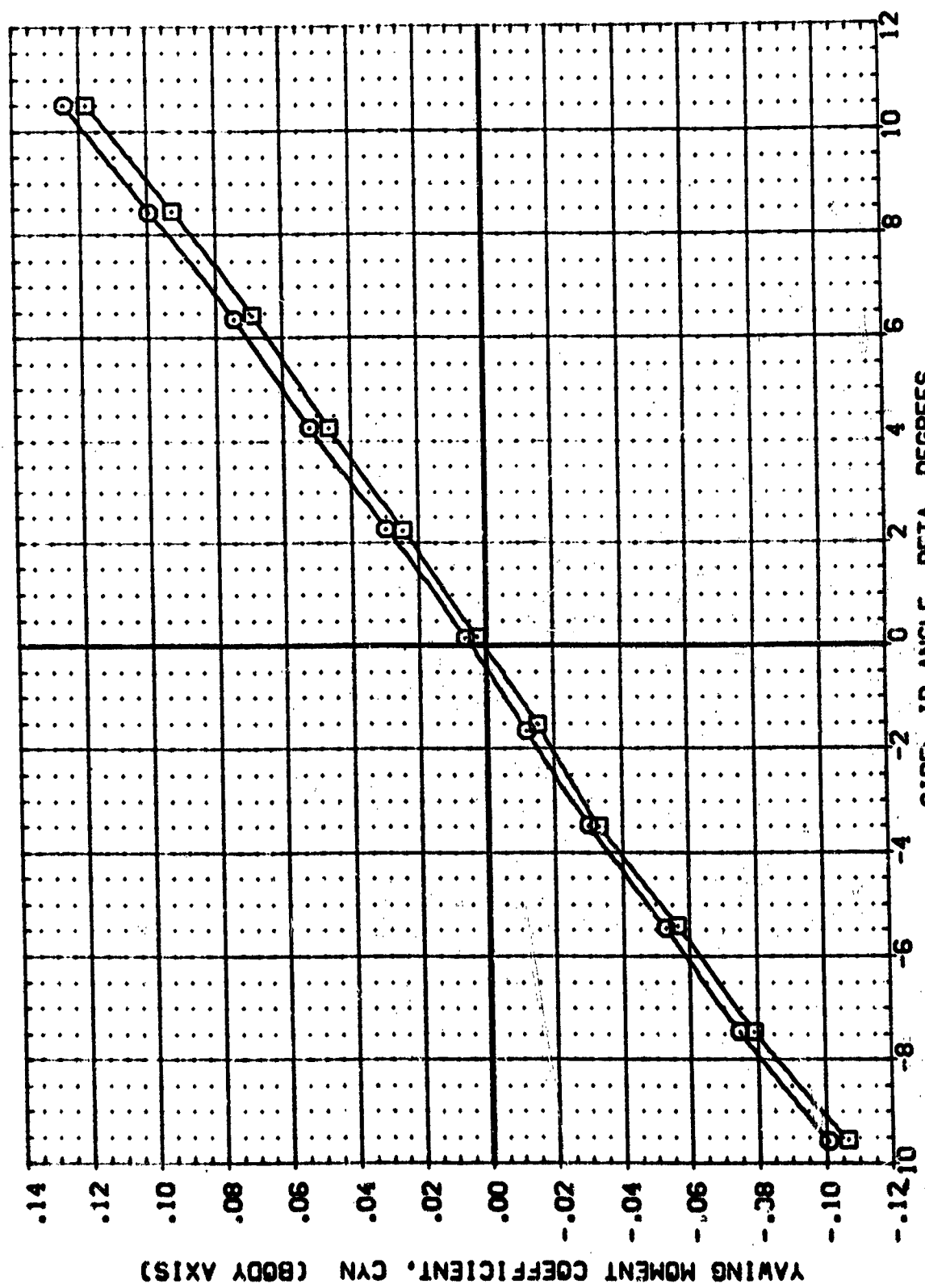


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBO (R87006) (R87007)  $\square$  CONFIGURATION DESCRIPTION AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF

REFERENCE INFORMATION  
 SIZE 2090.0000 IN.  
 LIFT 1200.0000 IN.  
 YPRP 930.0000 IN.  
 XPRP 1076.4800 IN.  
 YMRP 400.0000 IN.  
 ZMRP 0.0000 IN.  
 SCALE 400.0000 IN.

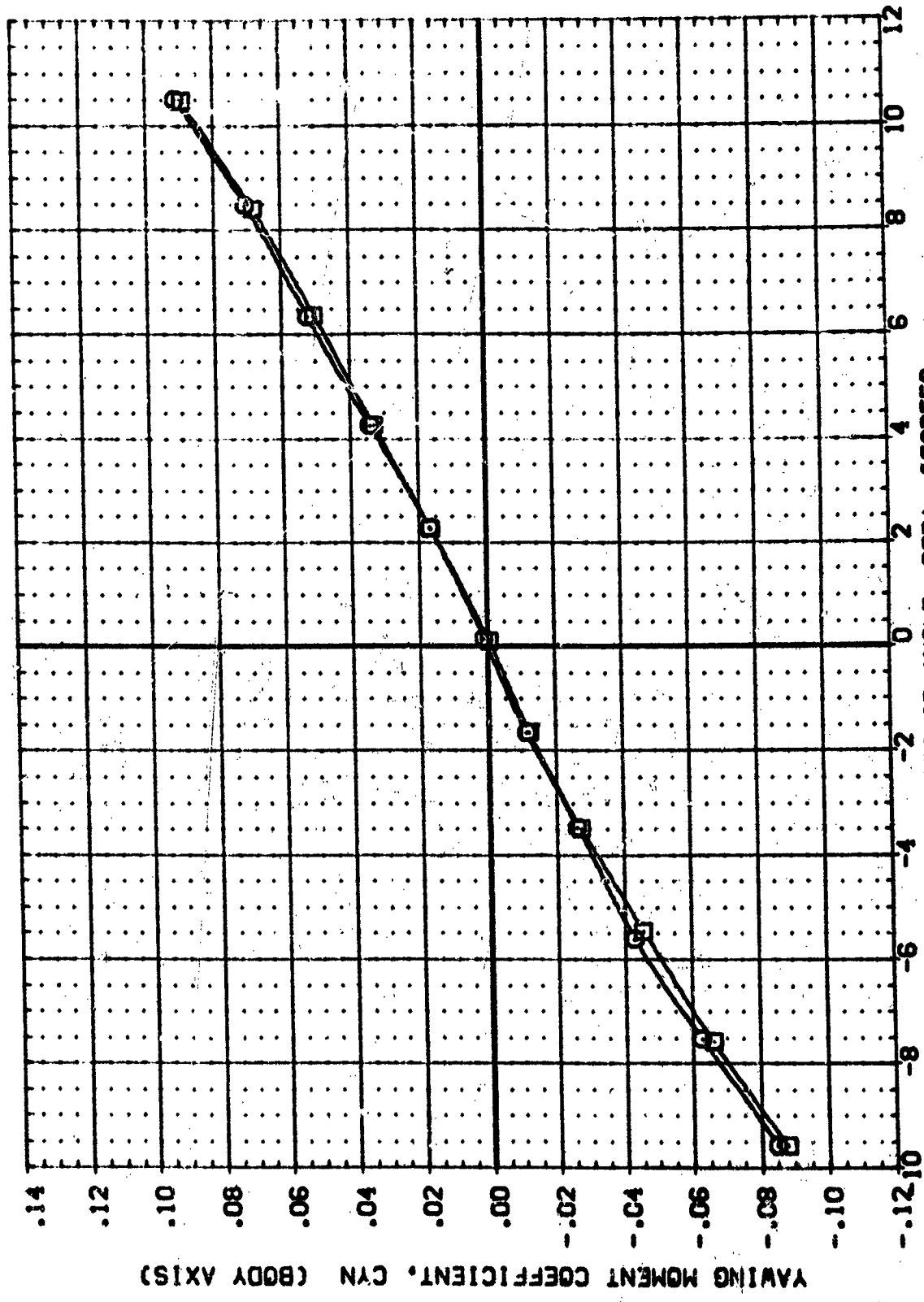


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.  
 (B)MACH = 7.32

DATA SET SYMBL. CONFIGURATION DESCRIPTION  
 (187006) 187006 09 T10 AT2 PLAVE OFF  
 (187007) 187007 09 T10 AT2 PLAVE OFF

ALPHA AILRON ELEVON RUDDER  
 .000 .000 .000 .000  
 .000 .000 .000 10.000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6900 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

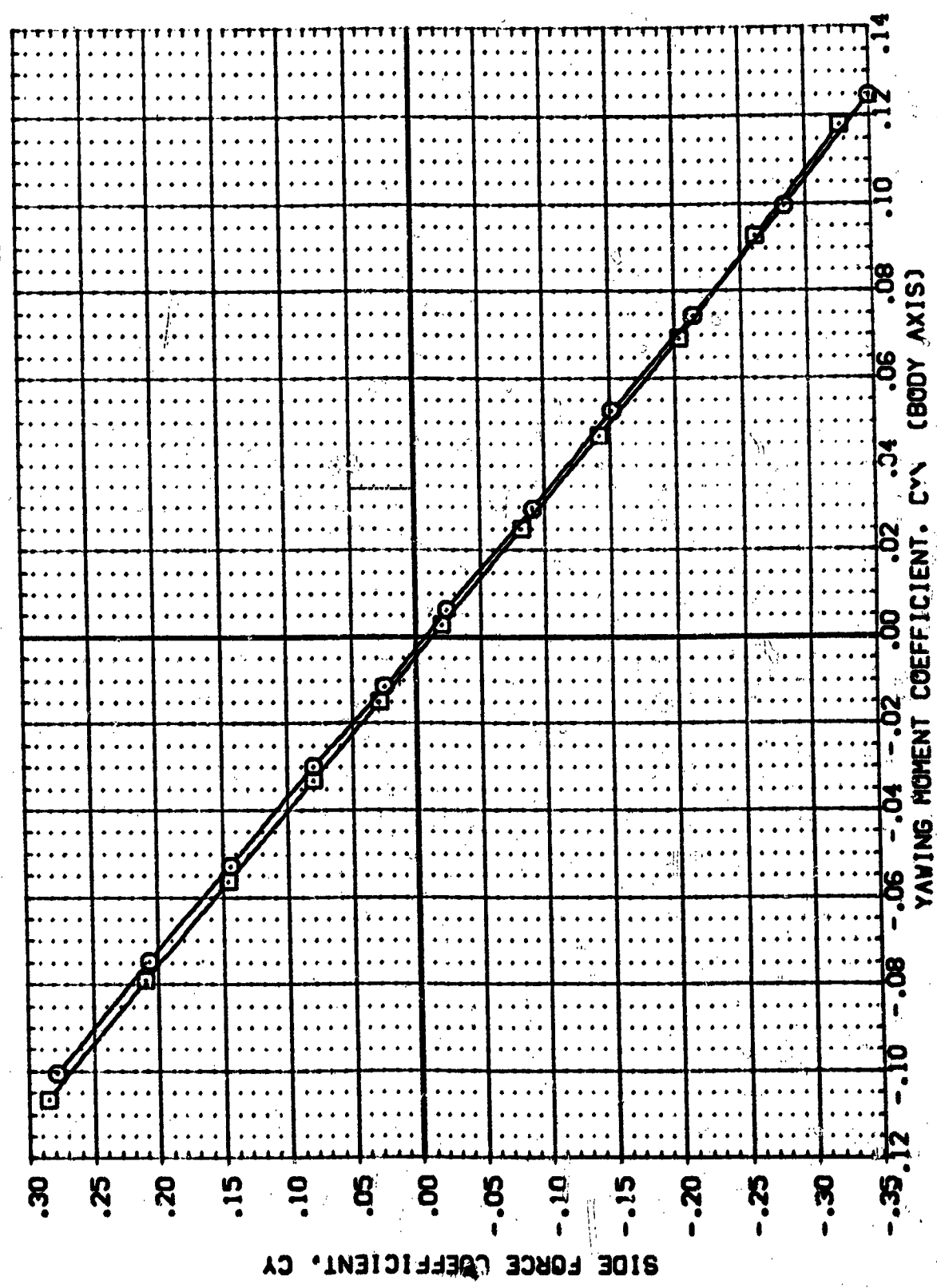


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBOL: (R87005) (R87007)

CONFIGURATION DESCRIPTION: AVES 3-5-169 IA10 CS T10 AT2 PLUVE OFF  
 AVES 3-5-169 IA10 CS T10 AT2 PLUVE OFF

REFERENCE INFORMATION:  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 536.6600 IN.  
 XWRP 1076.4800 IN.  
 YWRP .0000 IN.  
 ZWRP 400.0000 IN.  
 SCALE .0100

ALPHA .000  
 AILTRON .000  
 ELEVON .000  
 RUDDER .000  
 10.000

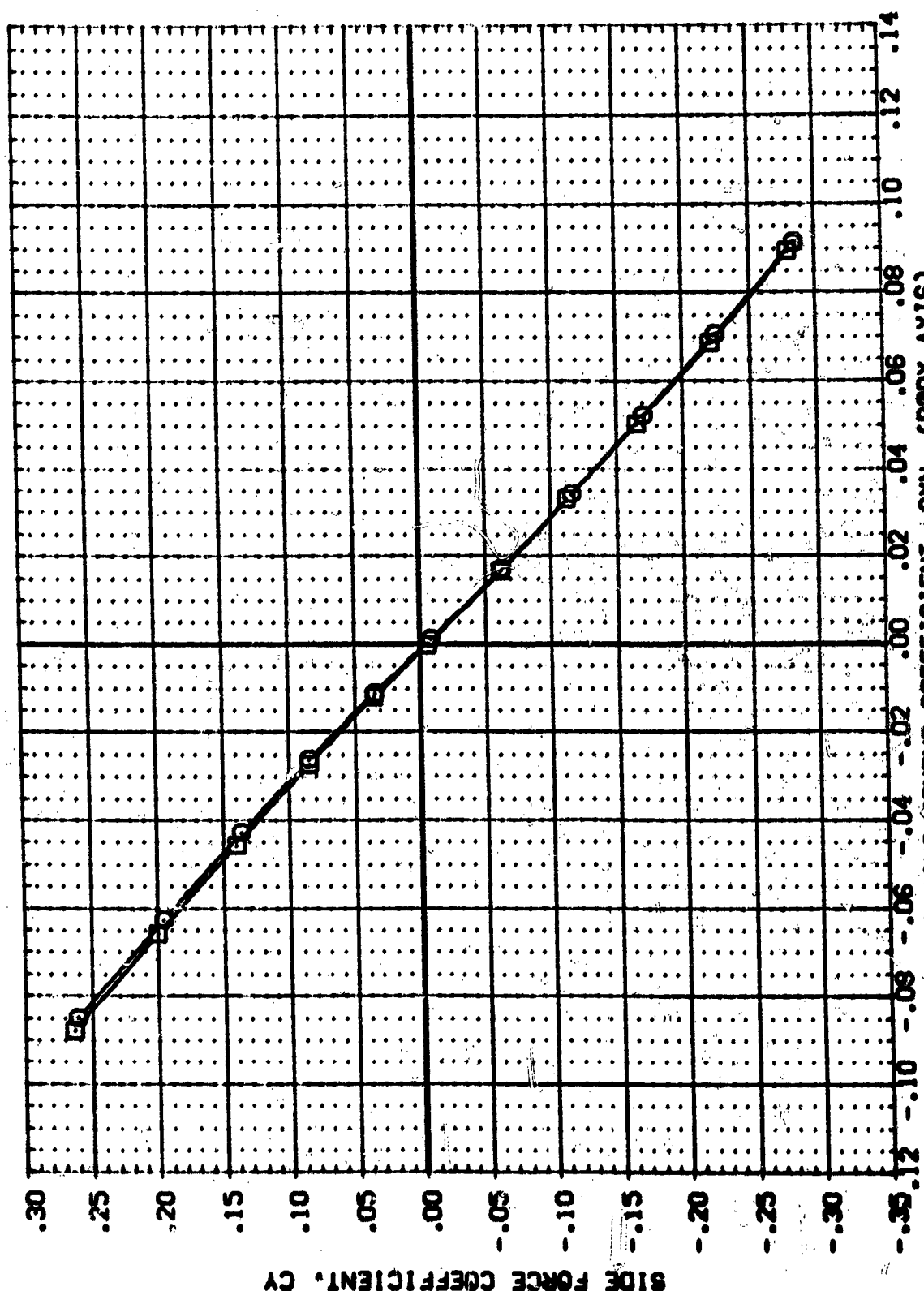


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL: (107001) (107002) (107003) (107004) (107005)

CONFIGURATION DESCRIPTION: ARES 3.5-169 IA10 OS T10 AT2 PLUVE OFF  
 ARES 3.5-169 IA10 OS T10 AT2 PLUVE OFF  
 DATA NOT AVAILABLE  
 DATA NOT AVAILABLE

BETA: .000  
 .000  
 5.000  
 5.000

AIRION: .000  
 .000  
 .000  
 .000

ELEVON: .000  
 .000  
 .000  
 .000

RUDDER: .000  
 .000  
 .000  
 .000

REFERENCE INFORMATION:  
 SREF: 2650.0000 SO.FT.  
 LREF: 1250.0000 IN.  
 BREF: 936.6800 IN.  
 XREF: 1076.4800 IN.  
 YREF: 400.0000 IN.  
 ZREF: 400.0000 IN.  
 SCALE: .0100

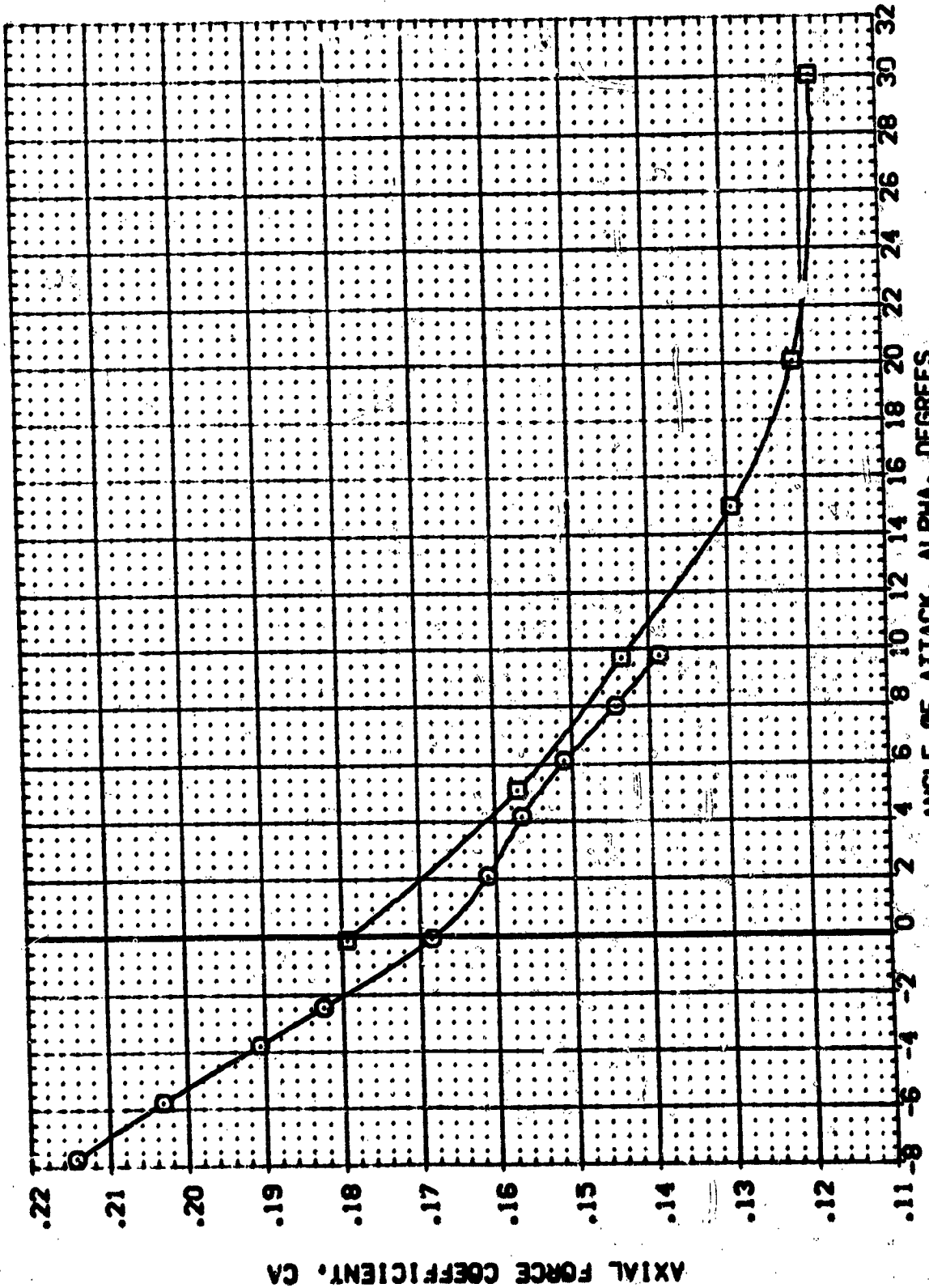


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26





DATA SET SYMBO. CONFIGURATION DESCRIPTION  
 (1807011) APES 3.5-169 IAI0 CS T10 AT2 PLUVE OFF  
 (1807002) APES 3.5-169 IAI0 CS T10 AT2 PLUVE OFF  
 (1807004) DATA NOT AVAILABLE  
 (1807003) DATA NOT AVAILABLE

BETA .000  
 .000  
 5.000  
 5.000

AILRON .000  
 .000  
 .000  
 .000

ELEVON .000  
 .000  
 .000  
 .000

SLIDER .000  
 .000  
 .000  
 .000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BRFP 936.6800 IN.  
 YWRP 1076.4800 IN.  
 ZWRP 400.0000 IN.  
 SCALE .0100

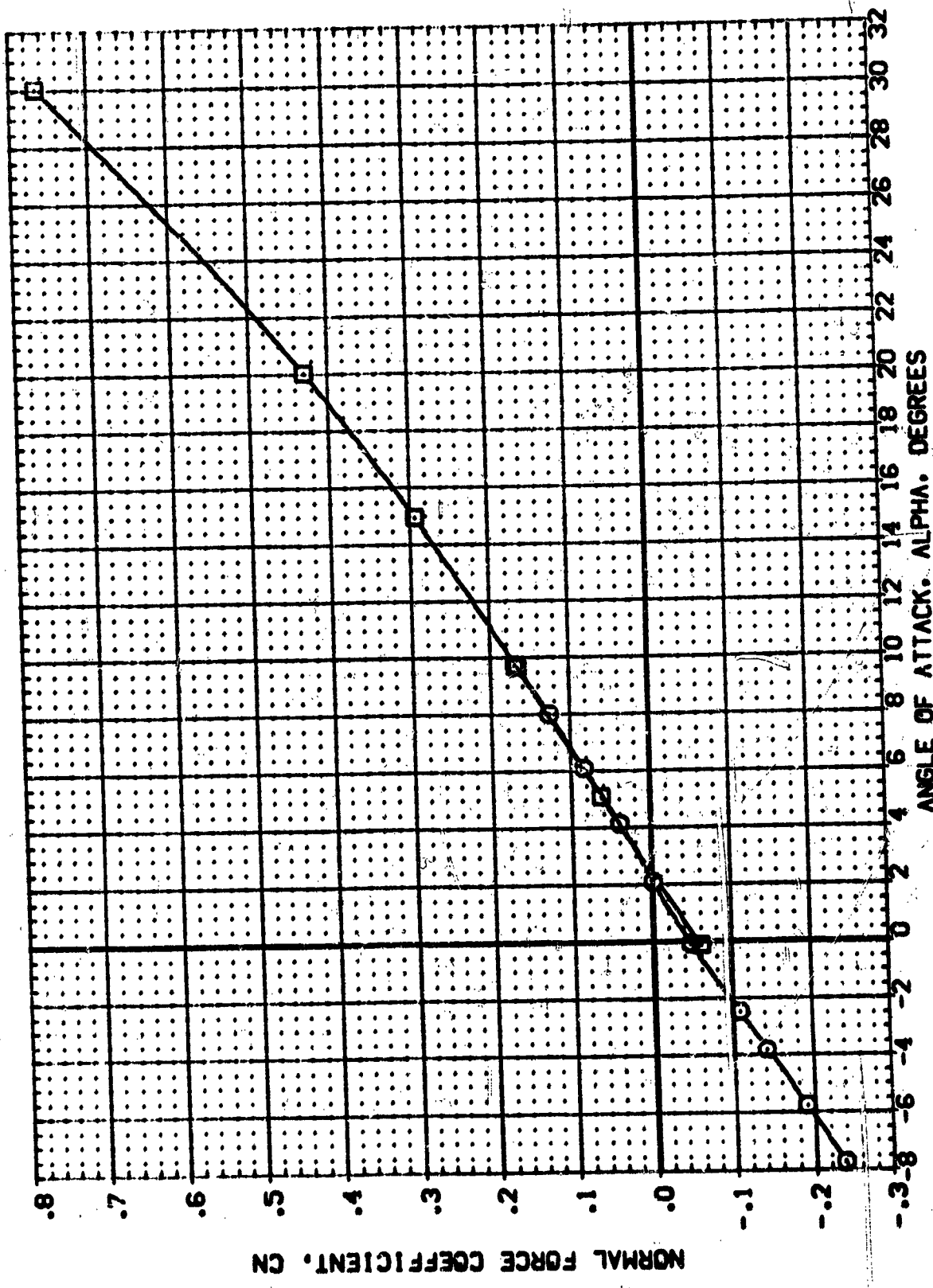


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.  
 (A)MACH = .26

DATA SET SYMB. CONFIGURATION DESCRIPTION

(C87011) DATA NOT AVAILABLE

(S87022) DATA NOT AVAILABLE

(R87034) ANES 3-5-169 IA10 OS T10 AT2 PLUVE OFF

(S87033) ANES 3-5-169 IA10 OS T10 AT2 PLUVE OFF

BETA .000

.000

5.000

5.000

5.000

AILRON .000

.000

.000

.000

.000

ELEVON .000

.000

.000

.000

.000

RUDDER .000

.000

.000

.000

.000

REFERENCE INFORMATION

SREF 2690.0000 SO.FT.

LREF 1290.0000 IN.

BREF 936.6800 IN.

XMRP 1076.4800 IN.

YMRP 400.0000 IN.

ZMRP .0100

SCALE

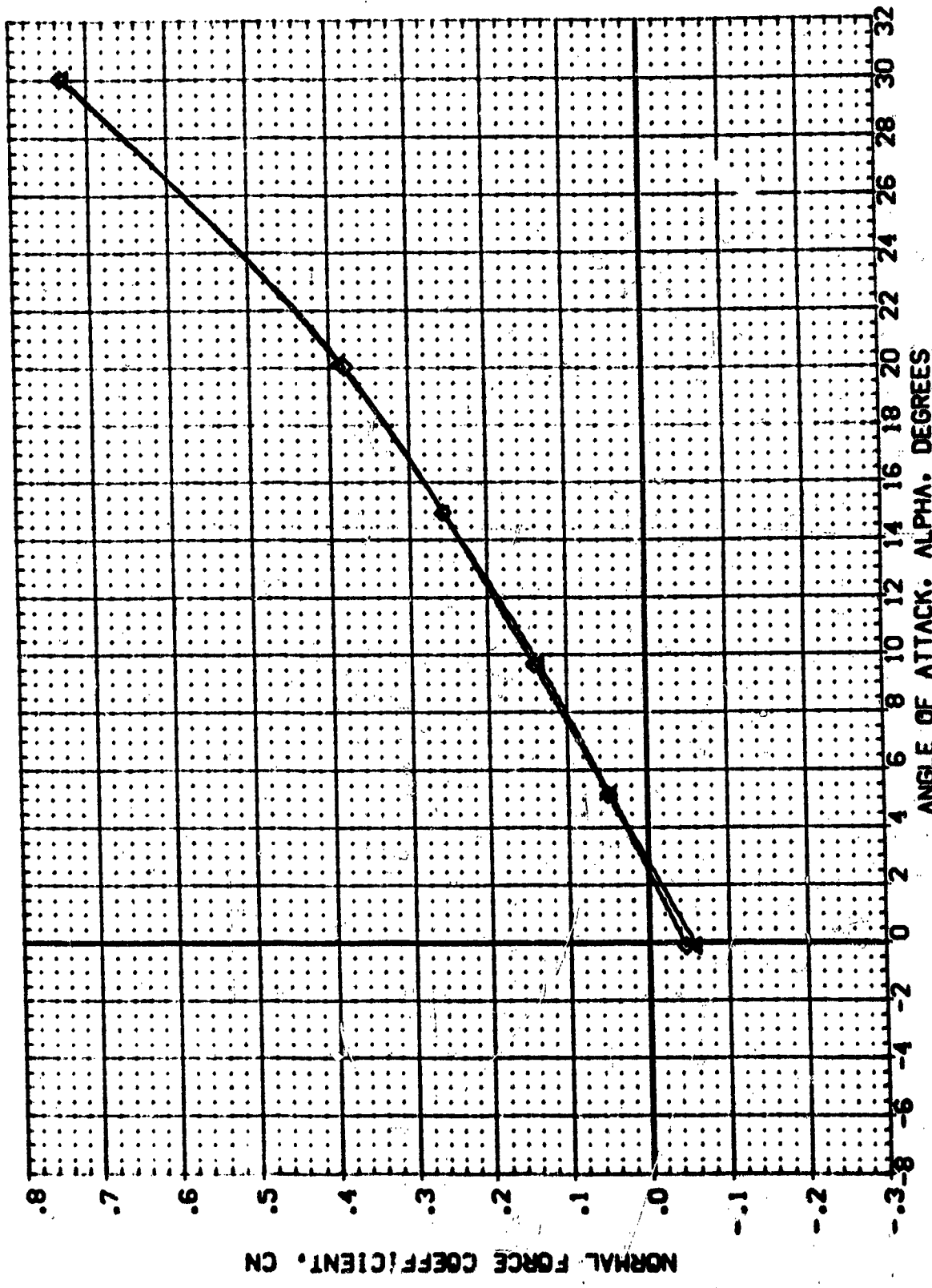


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBO. CONFIGURATION DESCRIPTION  
 (087011) AVES 3-5-169 JA10 OS T10 AT2 PLUVE OFF  
 (587002) AVES 3-5-169 JA10 OS T10 AT2 PLUVE OFF  
 (587004) DATA NOT AVAILABLE  
 (587003) DATA NOT AVAILABLE

BETA .000  
 .000  
 5.000  
 5.000

AILRON .000  
 .000  
 .000  
 .000

ELEVON .000  
 .000  
 .000  
 .000

RUDDER .000  
 .000  
 .000  
 .000

REFERENCE INFORMATION

SREF	2690.0000	SO.FT.
LREF	1290.0000	IN.
BREF	936.6800	IN.
XMRP	1076.4800	IN.
YMRP	.0000	IN.
ZMRP	400.0000	IN.
SCALE	.0100	

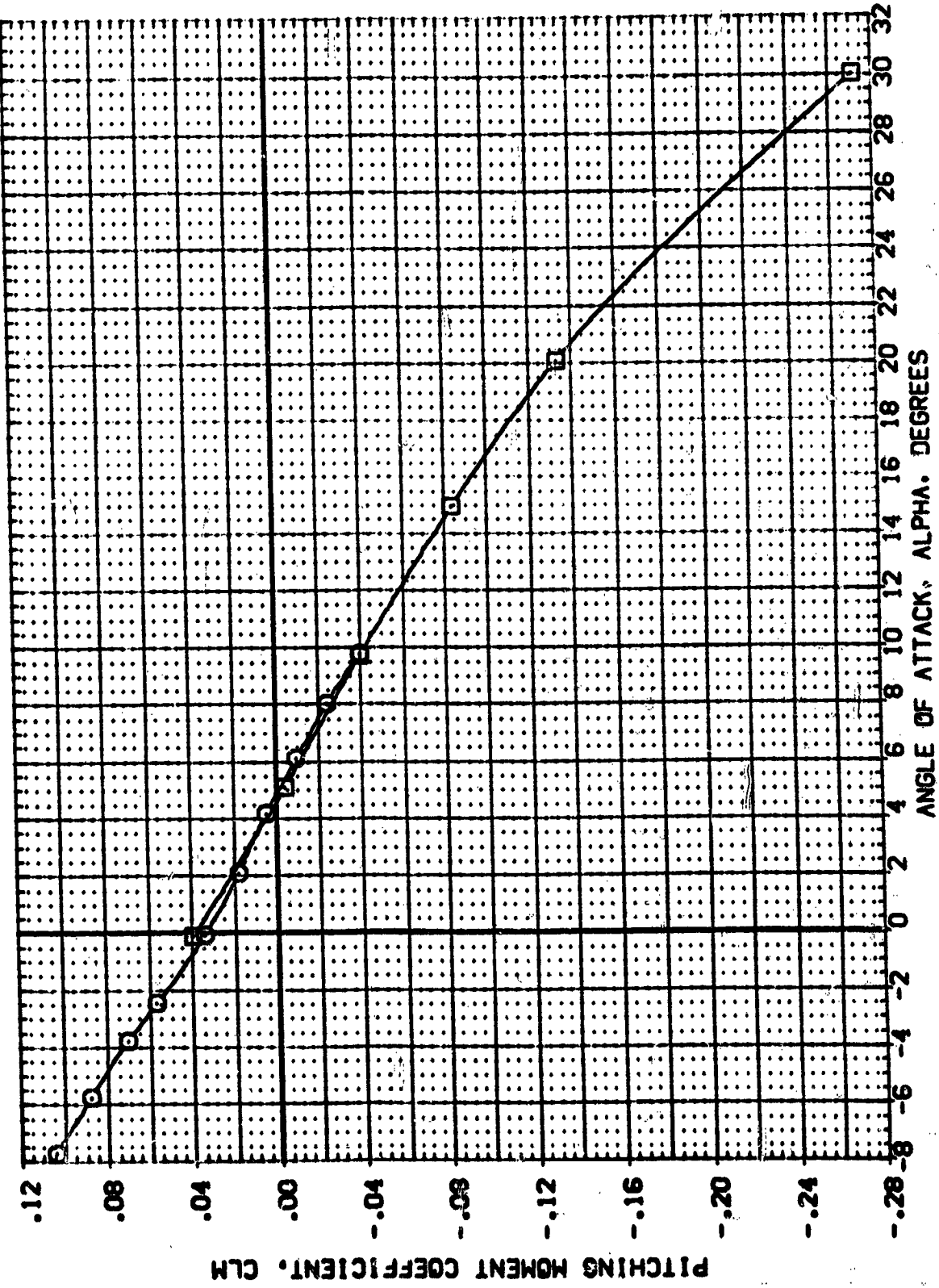


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBO. CONFIGURATION DESCRIPTION  
 (587011) DATA NOT AVAILABLE  
 (587022) DATA NOT AVAILABLE  
 (587033) AVES 3.5-169 IA10 OS T10 AT2 PLUVE OFF  
 (587033) AVES 3.5-169 IA10 OS T10 AT2 PLUVE OFF

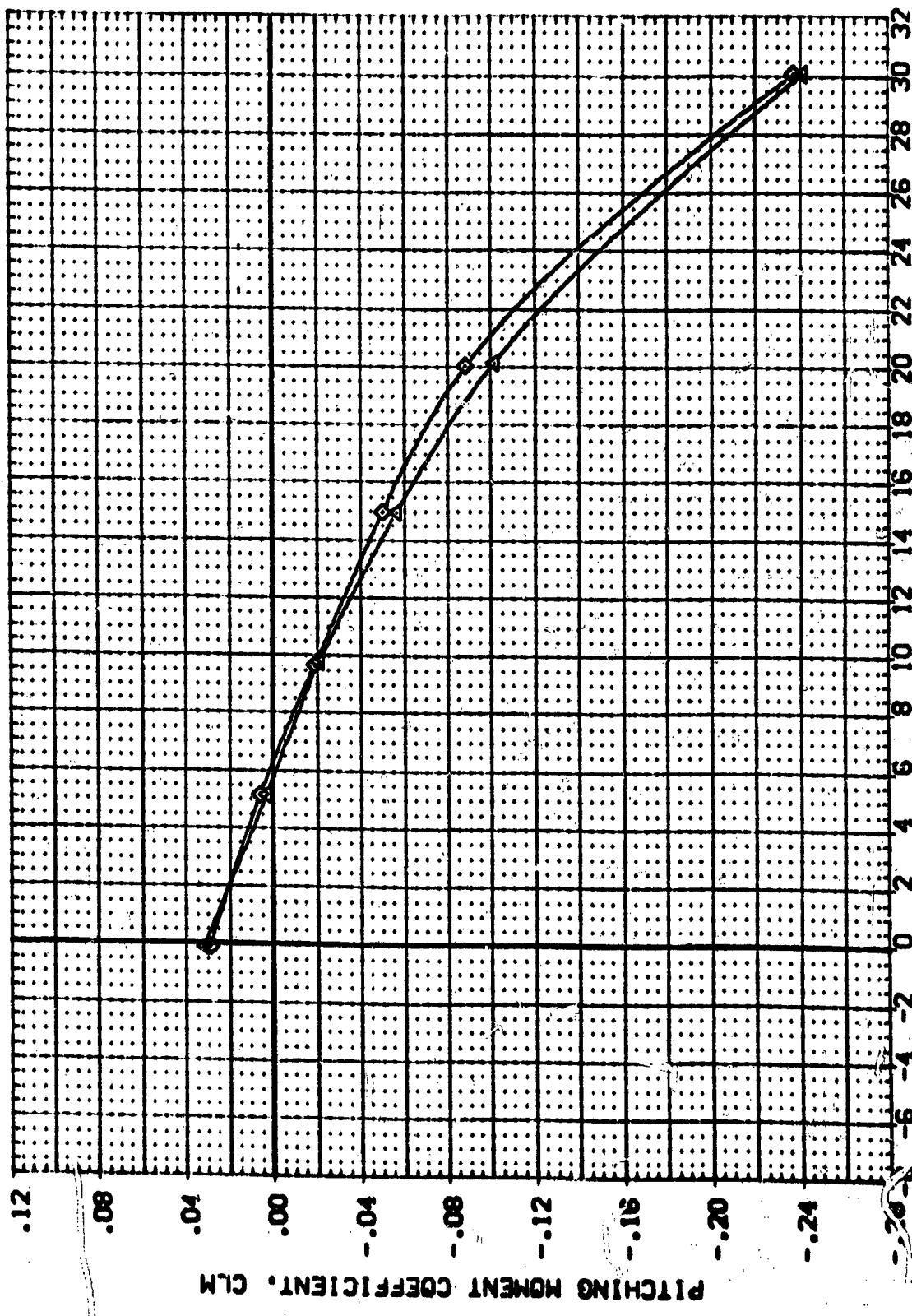
BETA .000  
 .000  
 5.000  
 5.000

AIRLON .000  
 .000  
 .000  
 .000

ELEVON .000  
 .000  
 .000  
 .000

RUDDER .000  
 .000  
 .000  
 .000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100



ANGLE OF ATTACK, ALPHA, DEGREES  
**FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.**  
 (B)MACH = 7.32

DATA SET SYMBOL: (187003)  
 CONFIGURATION DESCRIPTION: 09 T10 AT2 PLUVE OF  
 AMES 3-5-109 IA10 09 T10 AT2 PLUVE OF  
 AMES 3-5-109 IA10 09 T10 AT2 PLUVE OF  
 DATA NOT AVAILABLE  
 DATA NOT AVAILABLE

BETA: 0.000  
 5.000  
 5.000

ALLISON: 0.000  
 0.000  
 0.000

ELEVON: 0.000  
 0.000  
 0.000

RUDDER: 0.000  
 0.000  
 0.000

REFERENCE INFORMATION: SQ. FT.  
 SREF: 2690.0000  
 LREF: 1290.0000  
 BREF: 936.6900  
 XMRP: 1076.4800  
 YMRP: 400.0000  
 ZMRP: 0.0100  
 SCALE: .0100

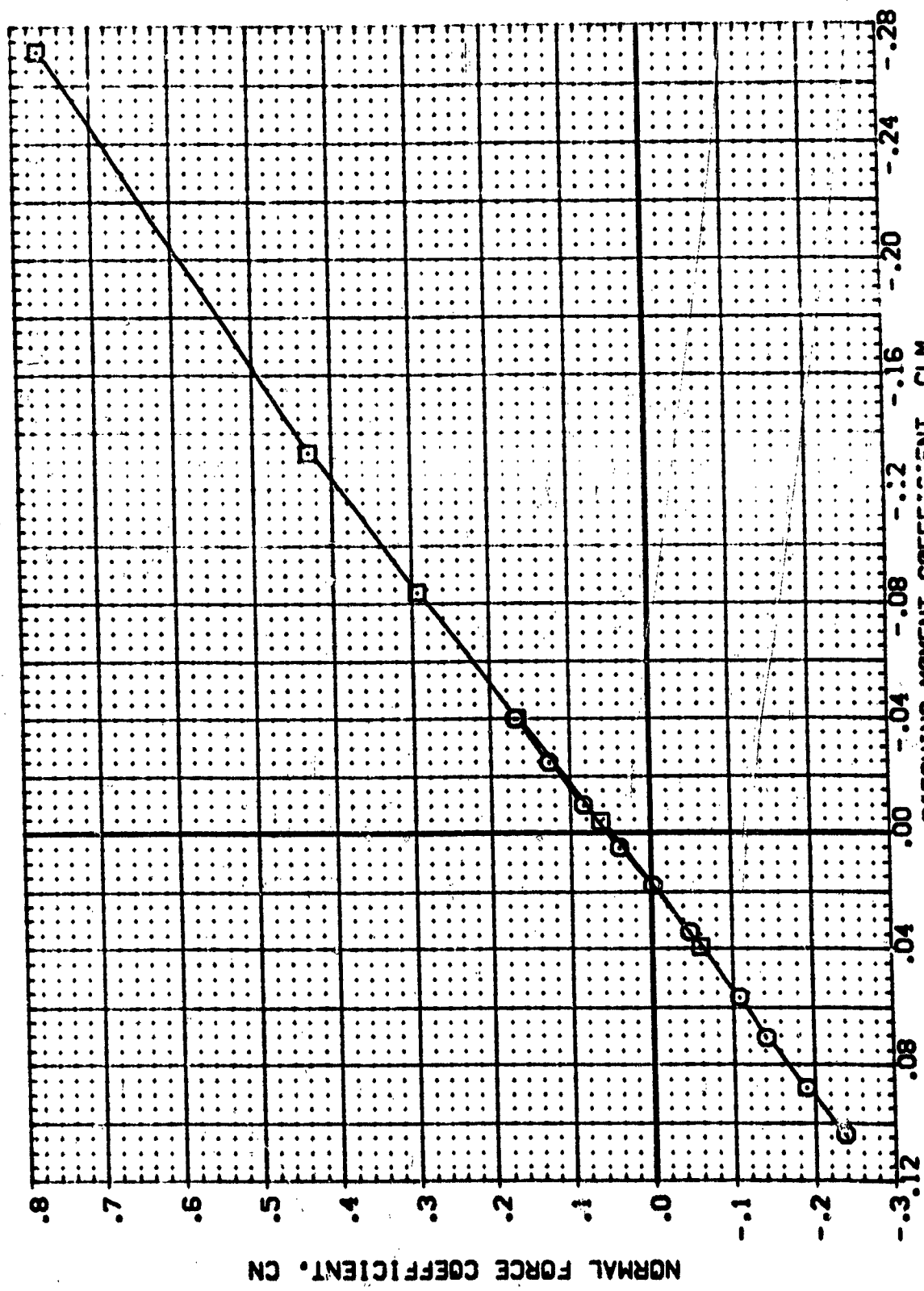


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(M)MACH = 5.26

DATA SET SYMBO: (DB7011) (SB7002) (RB7004) (RB7003)  
 CONFIGURATION DESCRIPTION: DATA NOT AVAILABLE, DATA NOT AVAILABLE, AVES 3.5-169 TA10, 09 T10 AT2 PLUVE OFF, 09 T10 AT2 PLUVE OFF  
 BETA: .000, 5.000, 5.000  
 AILRON: .000, .000, .000  
 ELEVON: .000, .000, .000  
 RUDDER: .000, .000, .000  
 REFERENCE INFORMATION: SREF 2690.0000 SO.FT., LREF 1290.0000 IN., BREF 936.5900 IN., XPRP 1076.4600 IN., YPRP .0000 IN., ZPRP 400.0000 IN., SCALE 400.0100

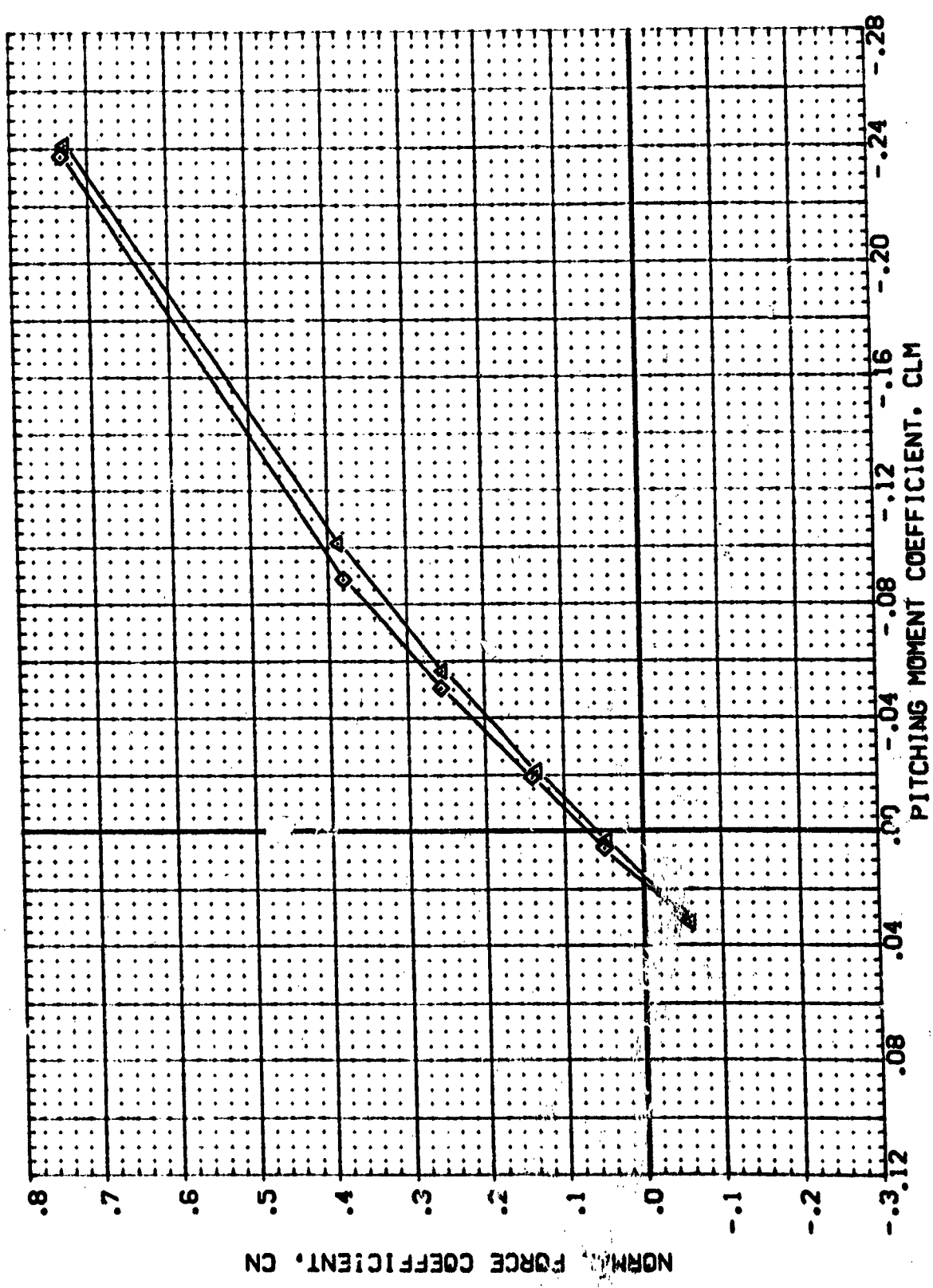


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBL. CONFIGURATION DESCRIPTION  
 (097011) ARES 3.5-189 IA10 OS T10 AT2 PLUVE OFF  
 (097002) ARES 3.5-189 IA10 OS T10 AT2 PLUVE OFF  
 (097004) DATA NOT AVAILABLE  
 (097003) DATA NOT AVAILABLE

BETA .000  
 .000  
 5.000  
 5.000

AIRRON .000  
 .000  
 .000  
 .000

ELEVON .000  
 .000  
 .000  
 .000

RUDDER .000  
 .000  
 .000  
 .000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

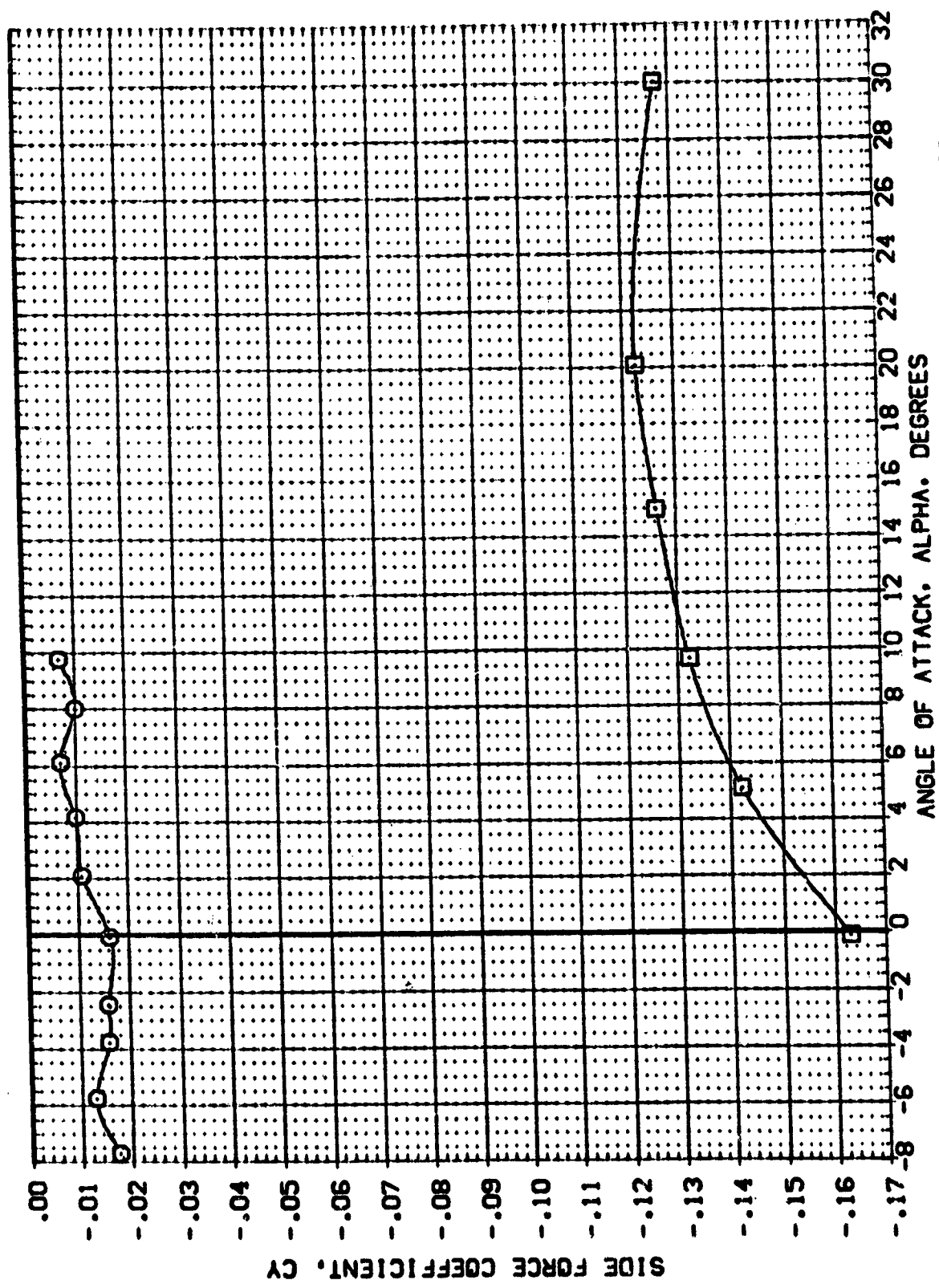


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26



DATA SET SYMBOL: (DB7011) DATA NOT AVAILABLE  
 (SB7022) DATA NOT AVAILABLE  
 (RB7004) AXES 3.5-169 I A10 OS T10 AT2 PLANE OFF  
 (RB7003) AXES 3.5-169 I A10 OS T10 AT2 PLANE OFF

BETA: .000  
 AIRBORN ELEVATION: .000  
 REFERENCE IN: 2690.0000 SQ. FT.  
 SREF: .000  
 LREF: .000  
 BREF: .000  
 XPRP: .000  
 YPRP: .000  
 ZPRP: .000  
 SCALE: .0100

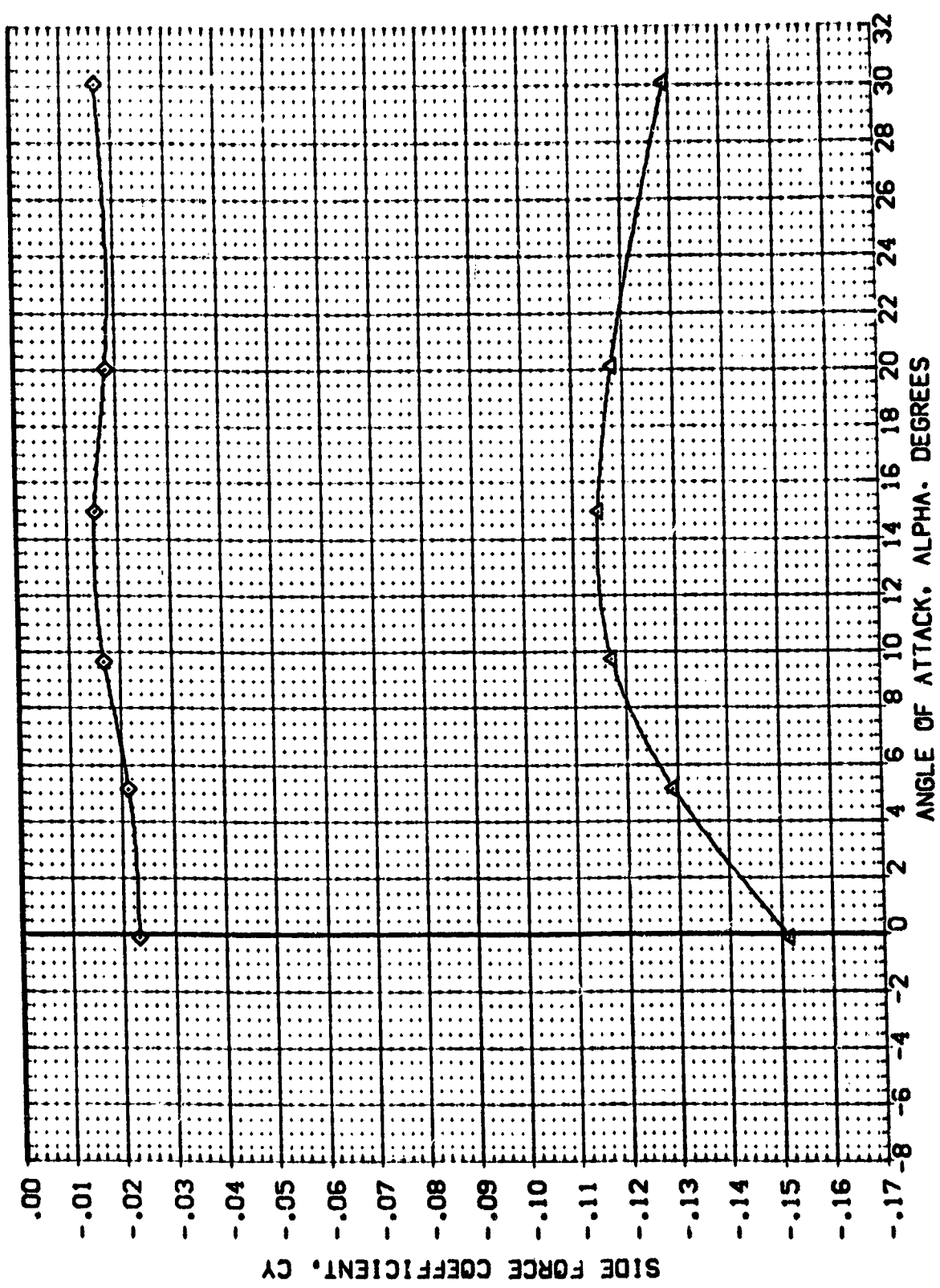


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL: (DB701) (SB7002) (RB7004) (RB7003)

CONFIGURATION DESCRIPTION: AXES 3-5-169 IAI0 2- T10 AT2 PLUVE OFF 3- T10 AT2 PLUVE OFF DATA NOT AVAILABLE DATA NOT AVAILABLE

BETA: .000  
5.000  
5.000

AILERON: .000  
.000  
.000

ELEVON: .000  
.000  
.000

RUDDER: .000  
.000  
.000

REFERENCE INFORMATION: SREF 7290.0000 IN. LREF 7290.0000 IN. BREF 536.6800 IN. XMRP 076.4800 IN. YMRP 0.0000 IN. ZMRP 0.0000 IN. SCALE .0125

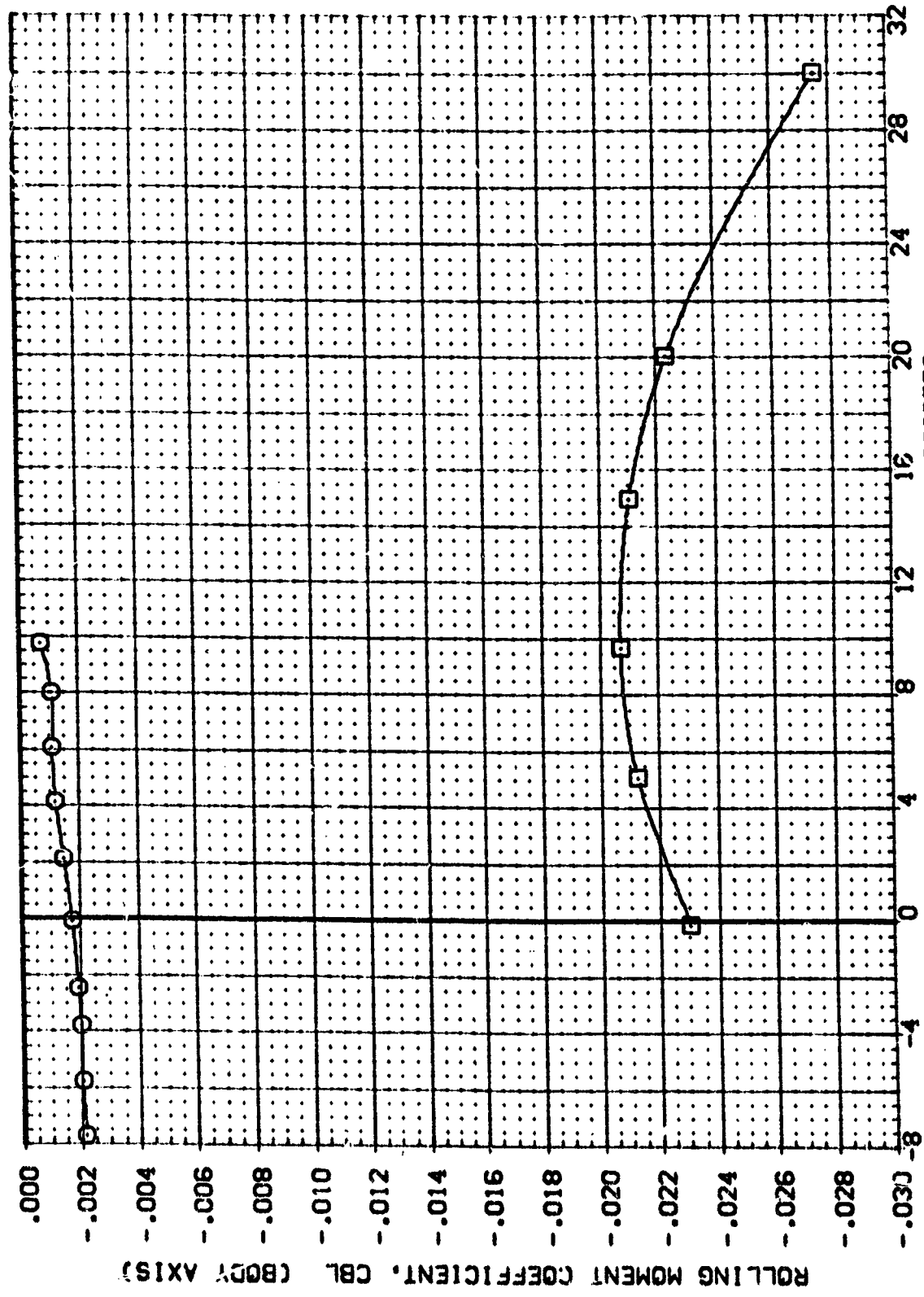


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBOL: (R57001) (R57002) (R57003) (R57004)

CONFIGURATION DESCRIPTION: DATA NOT AVAILABLE, DATA NOT AVAILABLE, AMES 3-5-169 IA10, 09 T10 AT2 FLUME OFF, 09 T10 AT2 FLUME OFF

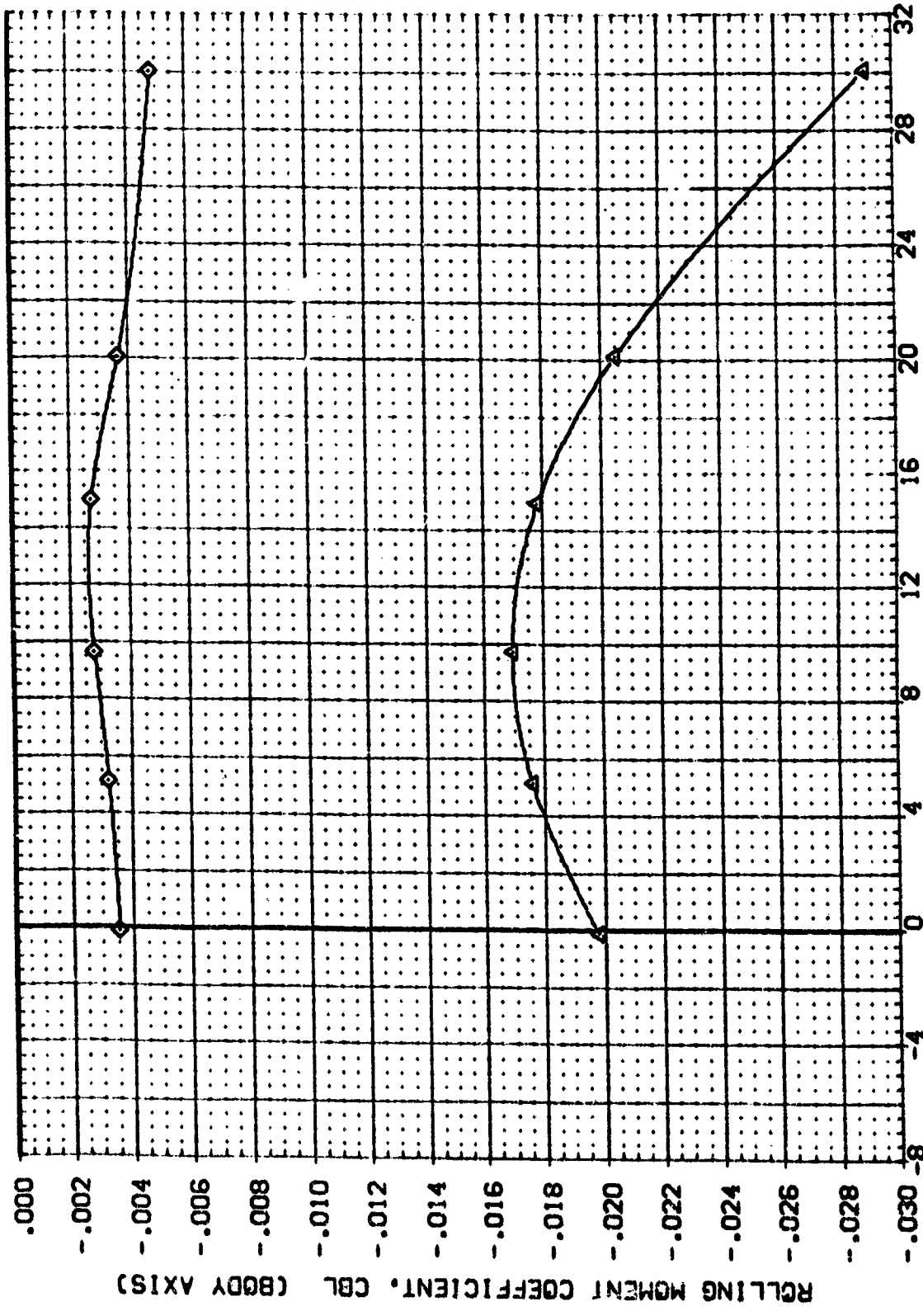
BETA: .000, 5.000, 5.000

AILERON: .000, .000, .000

ELEVON: .000, .000, .000

RUDDER: .000, .000, .000

REFERENCE INFORMATION: SREF 2690.0000 SO.FT. IN., LREF 1290.0000 IN., BRPF 936.6800 IN., XMRP 1076.4800 IN., YMRP .0000 IN., ZMRP 400.0000 IN., SCALE .0100



ANGLE OF ATTACK, ALPHA, DEGREES

FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(3)MACH = 7.32

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DATA SET SYMBL. CONFIGURATION DESCRIPTION  
 (027011) AMES 3-5-169 IA10 09 T10 AT2 PLUVE OFF  
 (587002) AMES 3-5-169 IA10 02 T10 AT2 PLUVE OFF  
 (887004) DATA NOT AVAILABLE  
 (887005) DATA NOT AVAILABLE

BETA AILRON ELEVON RUDDER REFERENCE INFORMATION SQ. FT.  
 .000 .000 .000 SREF 2690.0000  
 5.000 .000 .000 LREF 1290.0000  
 .000 .000 .000 BREF 936.6800  
 3.000 .000 .000 XMRP 1076.4800  
 YMRP 400.0000  
 ZMRP .0000  
 SCALE .0100

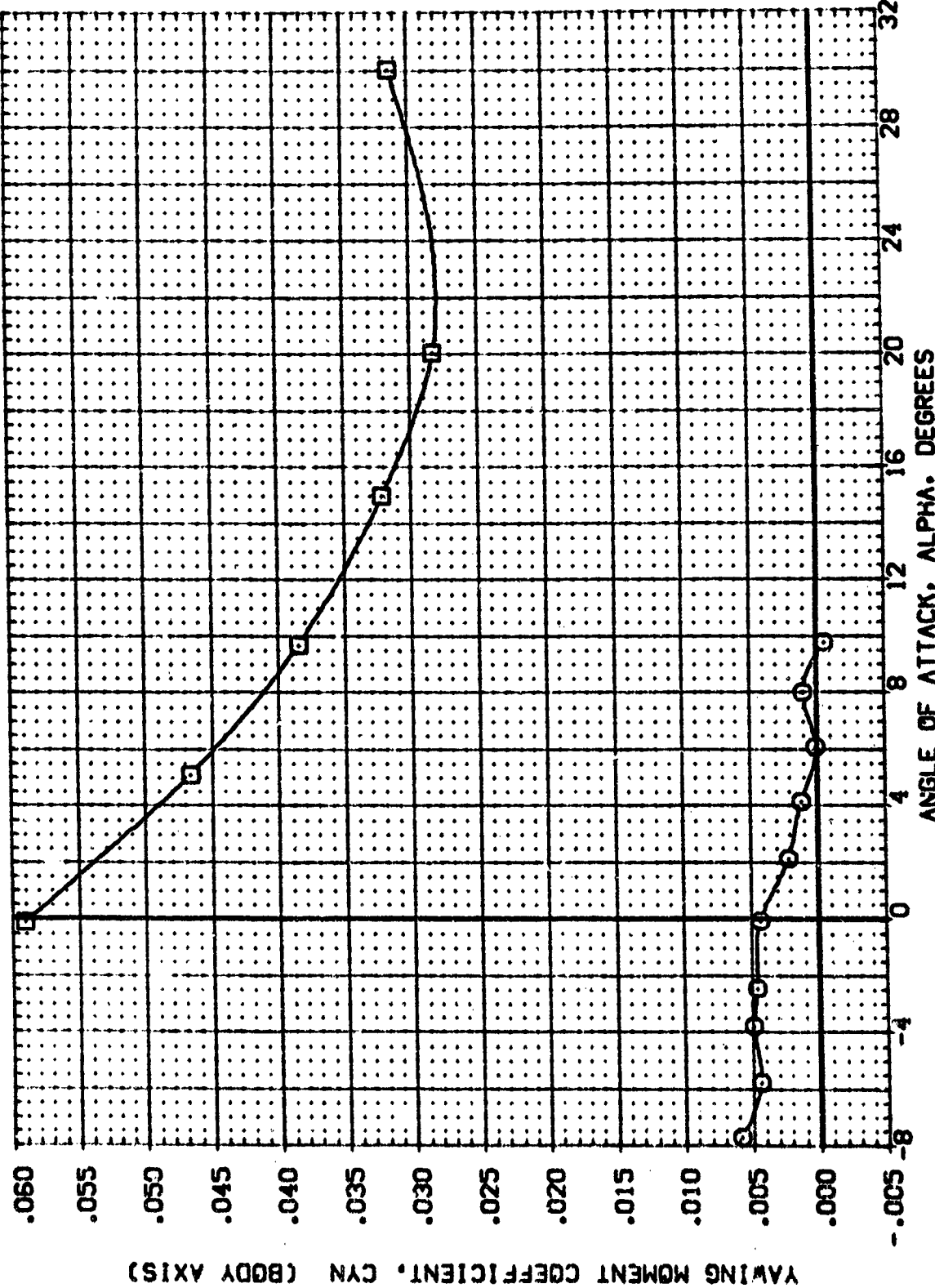


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A) MACH = 5.26

DATA SET SYMBOL: (R87011), (R87032), (R87074), (R87033)  
 CONFIGURATION DESCRIPTION: CS T10 AT2 PLUPE OFF, CS T10 AT2 PLUPE OFF, AVES 3.5-169 IA10  
 BETA: .000, 5.000, 5.000  
 AIRLON: .000, .000, .000  
 ELEVON: .000, .000, .000  
 FLDBER: .000, .000, .000  
 REFERENCE INFORMATION: SREF 2650.0000 SO.FT. IN., LREF 1250.0000 IN., BREF 936.6800 IN., XPROP 1076.4800 IN., YPROP .0000 IN., ZPROP 400.0000 IN., SCALE .0100

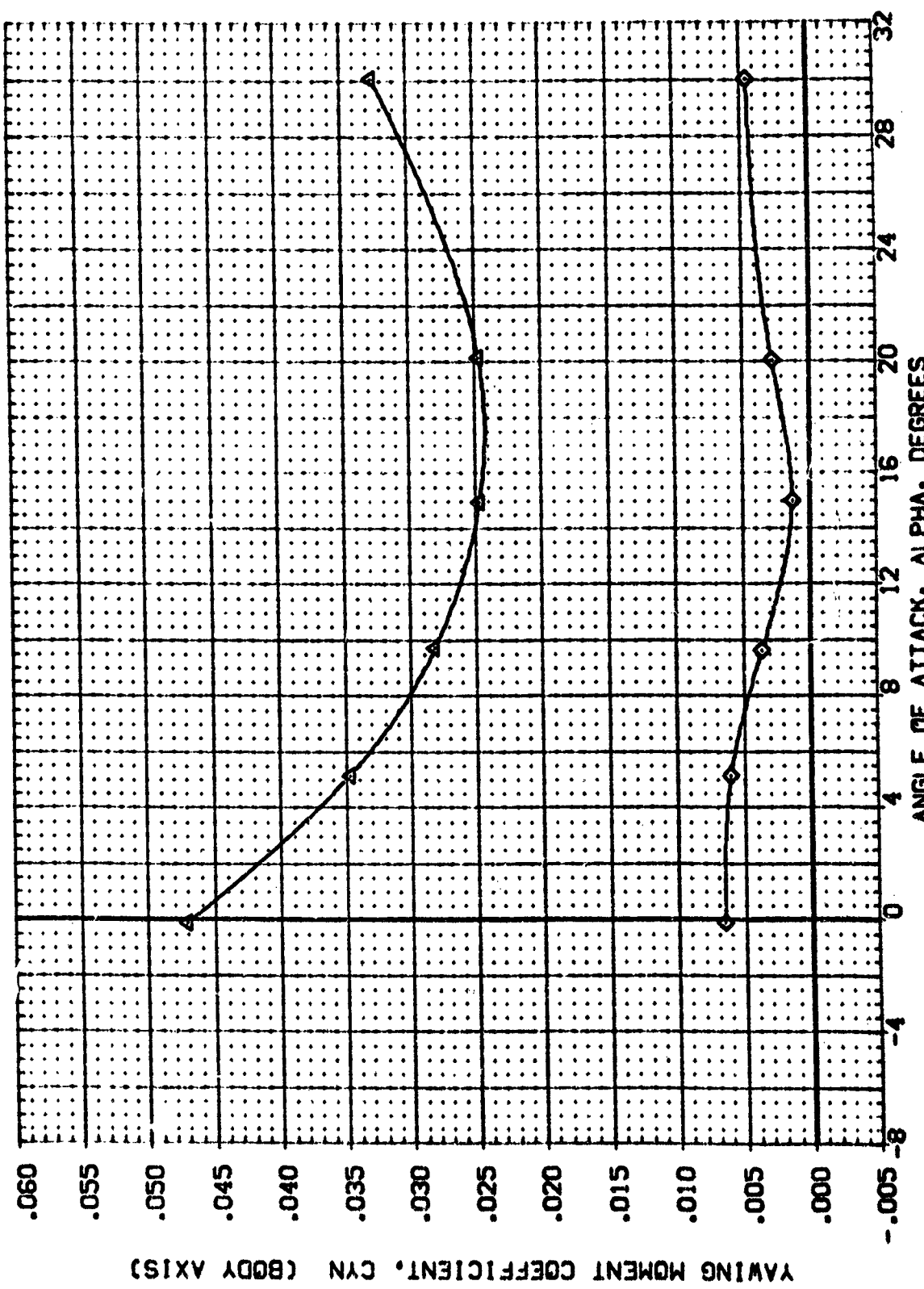


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMB. (1487008) (1487008)   
 CONFIGURATION DESCRIPTION (1487008) (1487008)   
 AYES 3.5-168 IA10 OS T10 AT2 PLUVE ON   
 AYES 3.5-168 IA10 OS T10 AT2 PLUVE ON

ALPHA .000 .000   
 AIRSON .000 .000   
 ELEVON .000 .000   
 RUDDER .000 10.000

REFERENCE INFORMATION   
 SREF 2690.0000 SQ.FT.   
 LREF 1290.0000 IN.   
 BREF 936.6800 IN.   
 XWRP 1076.4800 IN.   
 YWRP .0000 IN.   
 ZWRP 400.0000 IN.   
 SCALE .0100

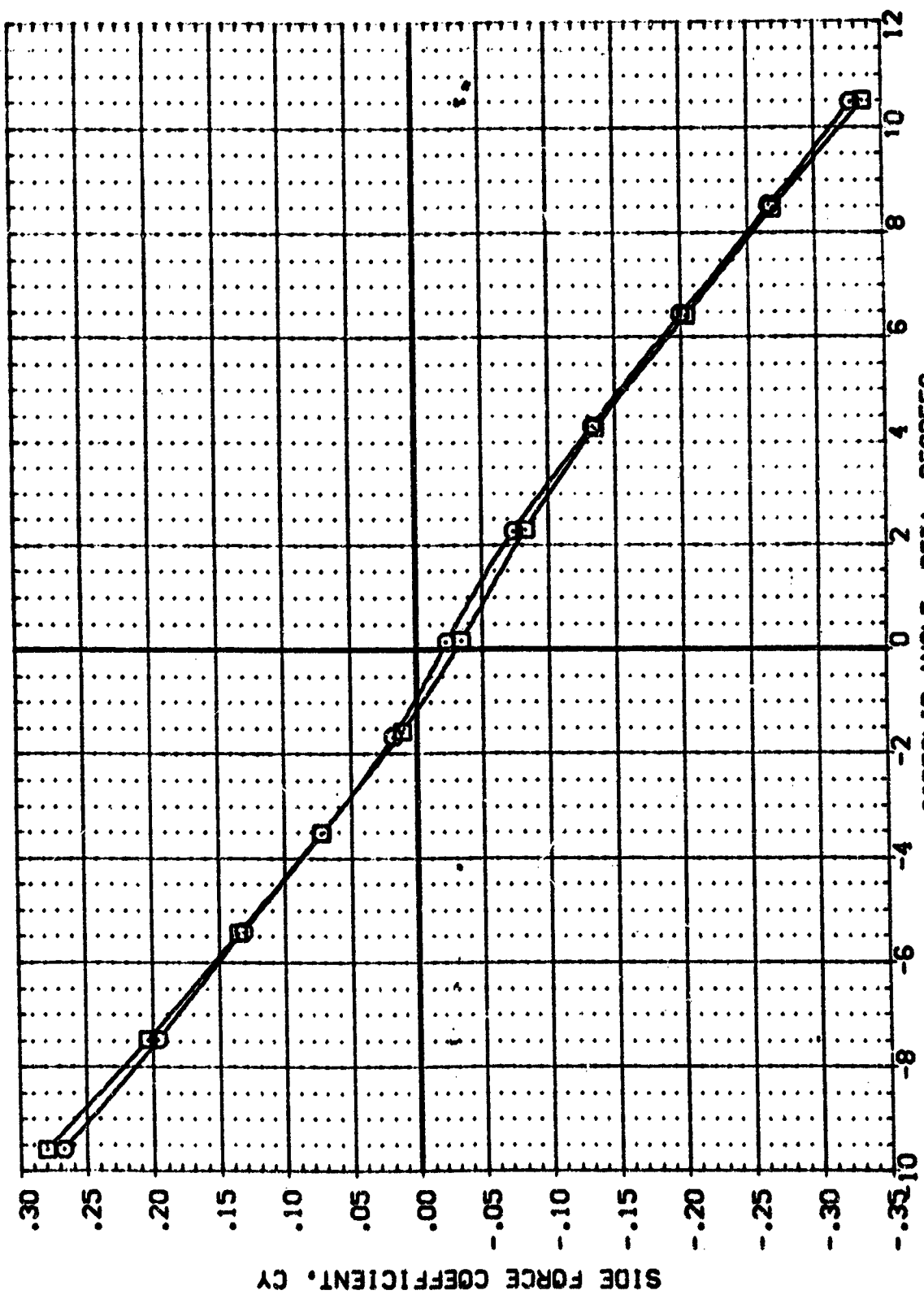


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBO. CONFIGURATION DESCRIPTION  
 (R87029) [ ] ARES 3.5-169 IA10 09 T10 AT2 PLUVE ON  
 (R87028) [ ] ARES 3.5-169 IA10 09 T10 AT2 PLUVE ON

REFERENCE INFORMATION  
 SREF 2690.0000 SO.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6900 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

ALPHA .000  
 AILRON .000  
 ELEVON .000  
 'RUDDER .000  
 10.000

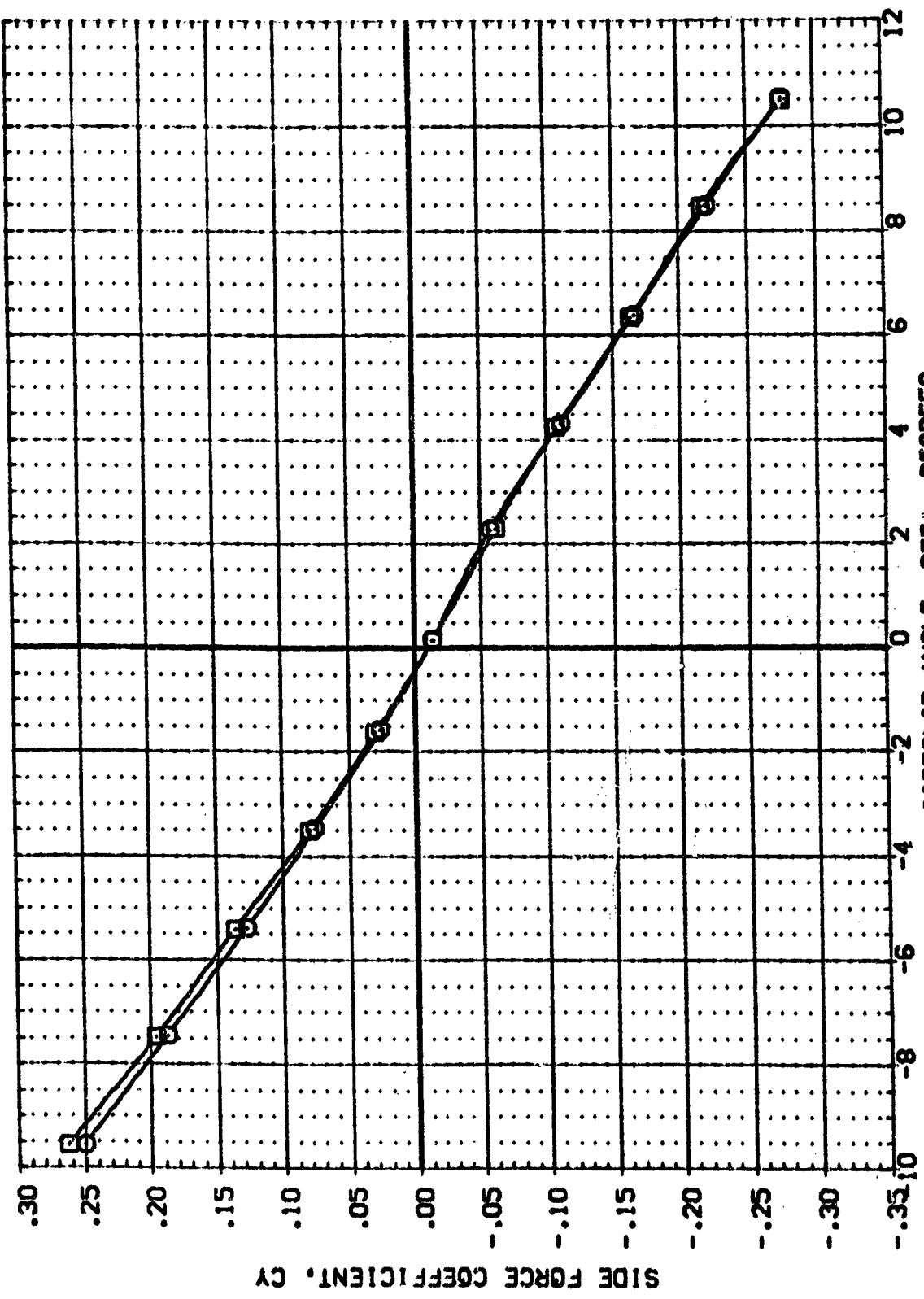


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (REFC09) AVES 3.5-169 IA10 C9 T10 AT2 PLUVE ON  
 (REFC08) AVES 3.5-169 IA10 C9 T10 AT2 PLUVE ON

ALPHA AILRON ELEVON RUDDER  
 .000 .000 .000 .000  
 .000 .000 .000 10.000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1250.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP 0.0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

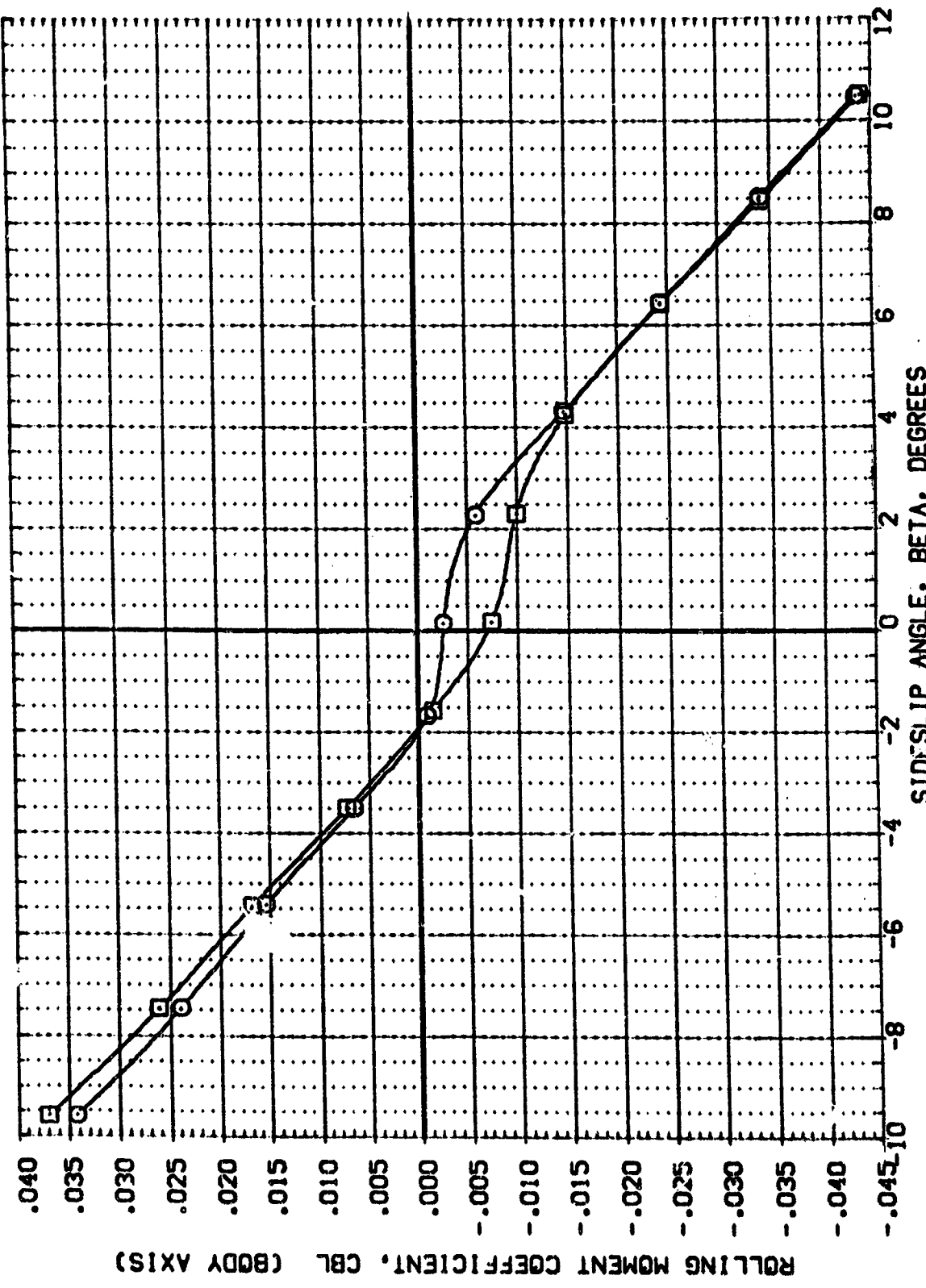


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26



DATA SET SYMBOL: (R97009) (R97009)  
 CONFIGURATION DESCRIPTION: AVES 3.5-169 IA10 CS TIC A12 PLANE ON AVES 3.5-169 IA10 CS TIC A12 PLANE ON  
 ALPHA: .000 .000  
 AIRLON: .000 .000  
 ELEVON: .000 .000  
 RUDDER: .000 .000  
 SCALE: 10.000  
 REFERENCE INFORMATION: SQ. FT. 2690.0000  
 SREF: 1290.0000 IN.  
 SREF: 936.6800 IN.  
 XMRP: 1076.4800 IN.  
 YMRP: .0000 IN.  
 ZMRP: 400.0000 IN.

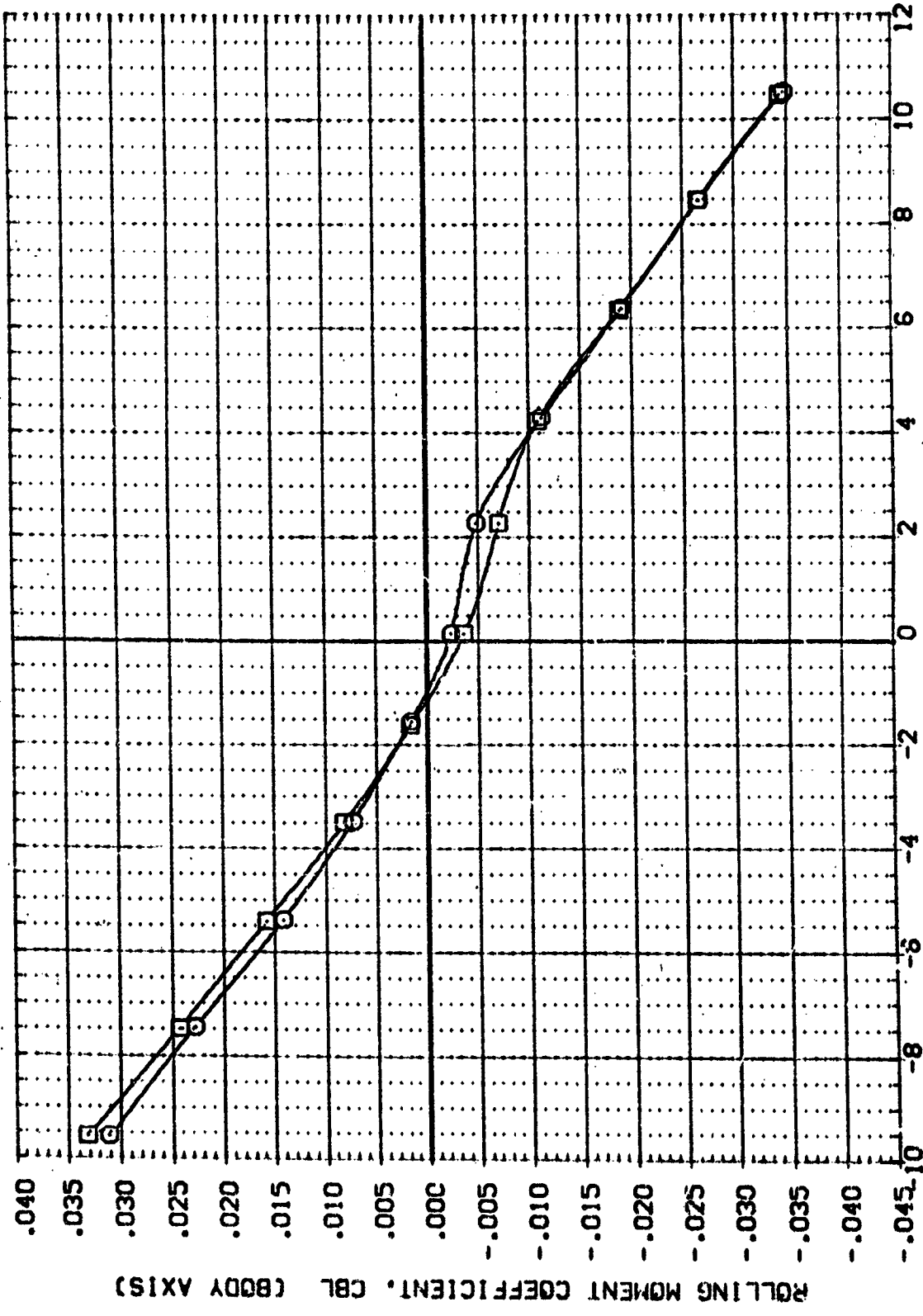


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.



DATA SET SYMBOL: (R87009) (R87008)  
 CONFIGURATION DESCRIPTION: AVES 3.5-169 IA10 08 T10 AT2 PLUVE ON / AVES 3.5-169 IA10 09 T10 AT2 PLUVE ON

REFERENCE INFORMATION:  
 SREF 2690.0000 SQ.FT.  
 LREF 1790.0000 IN.  
 BREF 536.6600 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE 400.0100

ALPHA: .000, .000  
 AIRLON: .000, .000  
 ELEVON: .000, .000  
 RUDDER: .000, 10.000

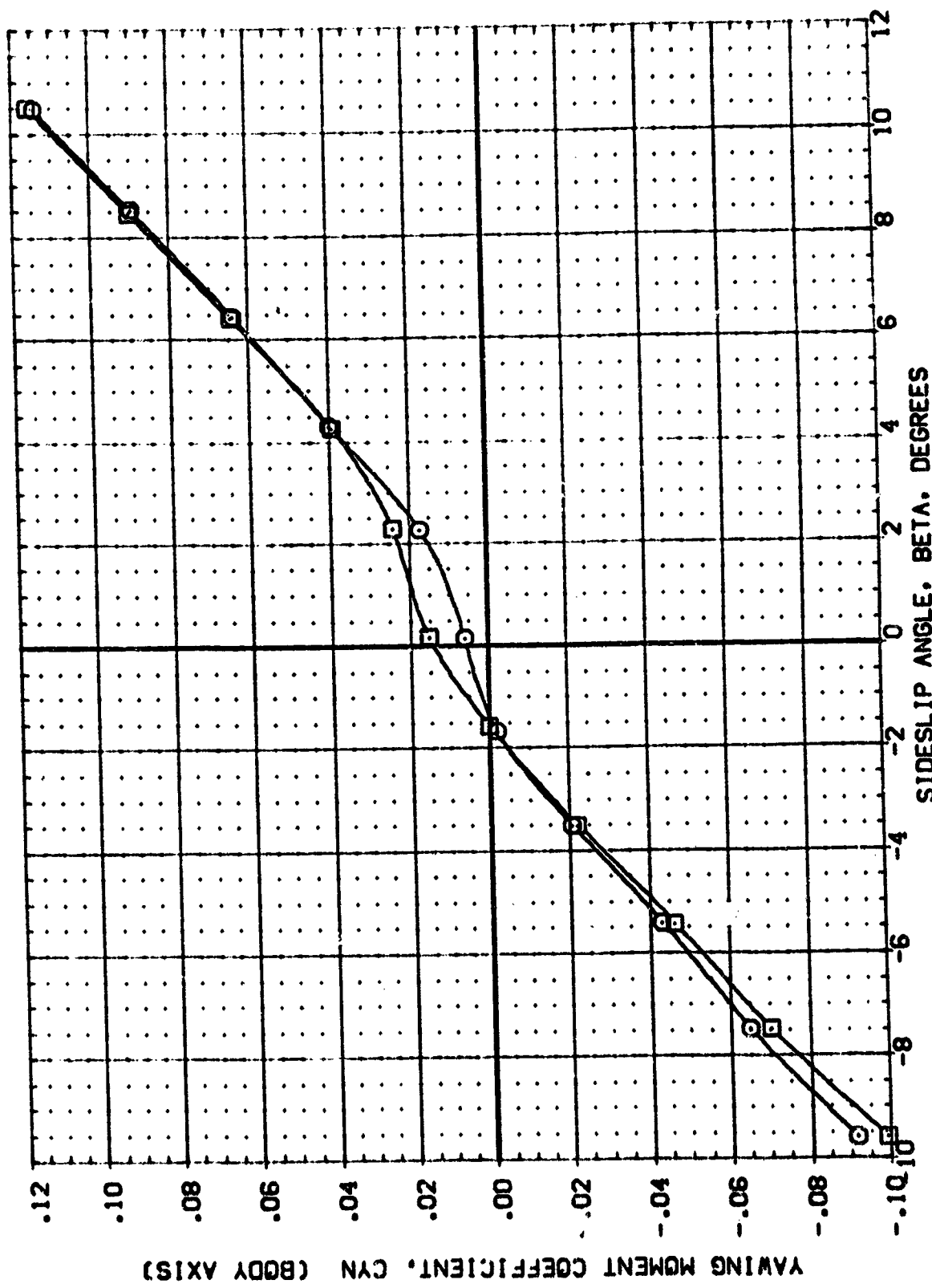


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBOL: (B77008) (R877008)

CONFIGURATION DESCRIPTION: AVES 3.5-169 IA10 09 T10 AT2 PLANE ON / AVES 3.5-169 IA10 09 T10 AT2 PLANE ON

REFERENCE INFORMATION: SREF 2590.0000 SQ.FT. / LREF 1250.0000 IN. / XMRP 936.6900 IN. / YMRP 1076.4800 IN. / ZMRP 0000 IN. / SCALE 400.0000

ALPHA: .000 / AILERON: .000 / ELEVON: .000 / RUDDER: .000 / 10.000

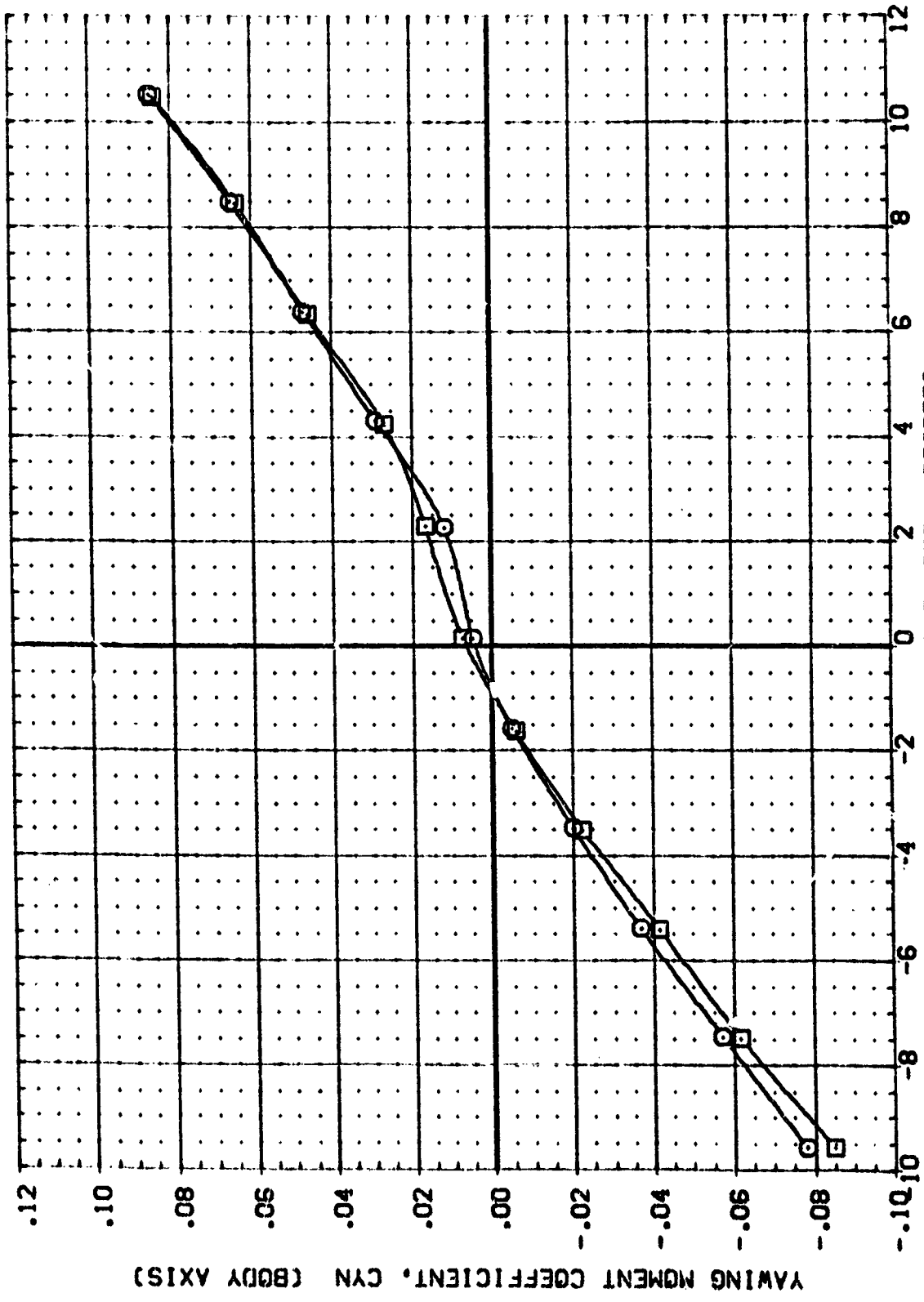


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (R67009) [ ] AVES 3-5-169 IA10 09 T10 AT2 PLUVE ON  
 (R67008) [ ] AVES 3-5-169 IA10 09 T10 AT2 PLUVE ON

ALPHA .000 .000  
 AITLON .000 .000  
 ELEVON .000 .000  
 RUDDER .000 10.000  
 REFERENCE INFORMATION  
 SREF 269C.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 536.8600 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

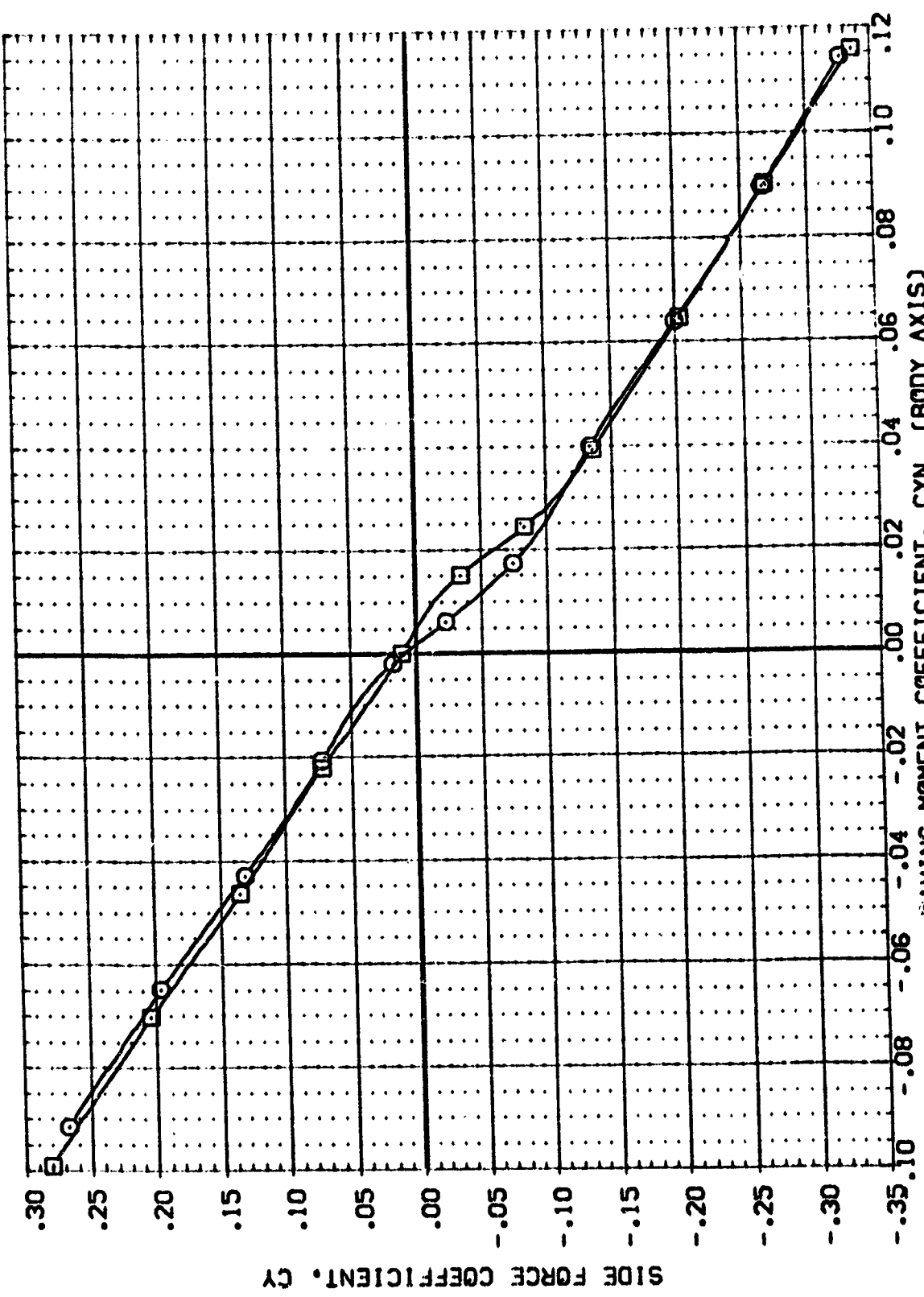


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBOL: (R87009) (R87008)  
 CONFIGURATION DESCRIPTION: AMES 3.5-169 (A10 OS T10 AT2 PLUPE ON) (R87008)  
 ALPHA: .000  
 AILERON: .000  
 ELEVON: .000  
 RUDDER: .000  
 REFERENCE INFORMATION:  
 SREF: 2690.0000 SQ.FT.  
 LREF: 1.290.0000 IN.  
 BREF: 5.36.6800 IN.  
 XMRP: 1076.4800 IN.  
 YMRP: .0000 IN.  
 ZMRP: 400.0000 IN.  
 SCALE: .0100

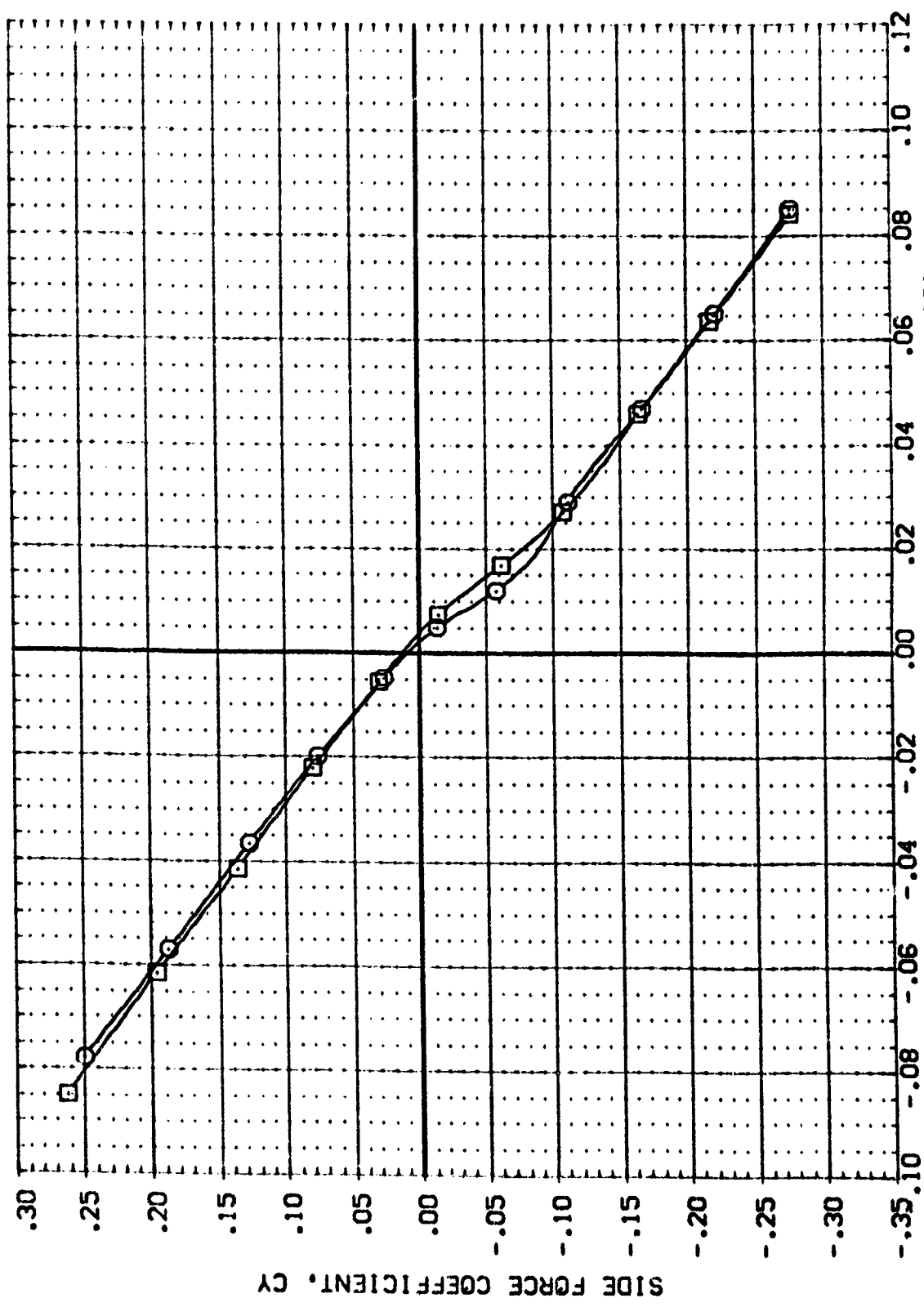


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.  
 (B)MACH = 7.32

LOCAL NORMAL FORCE COEFFICIENT DERIVATIVE WITH ALPHA, CN/A, PER DEGREE

DATA SET SYMBL. CONFIGURATION DESCRIPTION  
 (M87011) AVES 3.5-169 IA10 09 T10 AT2 PLUFE OFF  
 (L87005) AVES 3.5-169 IA10 09 T10 AT2 PLUFE OFF  
 (M87004) DATA NOT AVAILABLE  
 (V87005) DATA NOT AVAILABLE

BETA .000  
 .000  
 .000  
 .000

AILERON .000  
 .000  
 10.000  
 .000  
 10.000

ELEVON .000  
 .000  
 .000  
 .000  
 .000

RUDER .000  
 .000  
 .000  
 .000

REFERENCE INFORMATION  
 SREF 2690.0000 SO.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP 400.0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

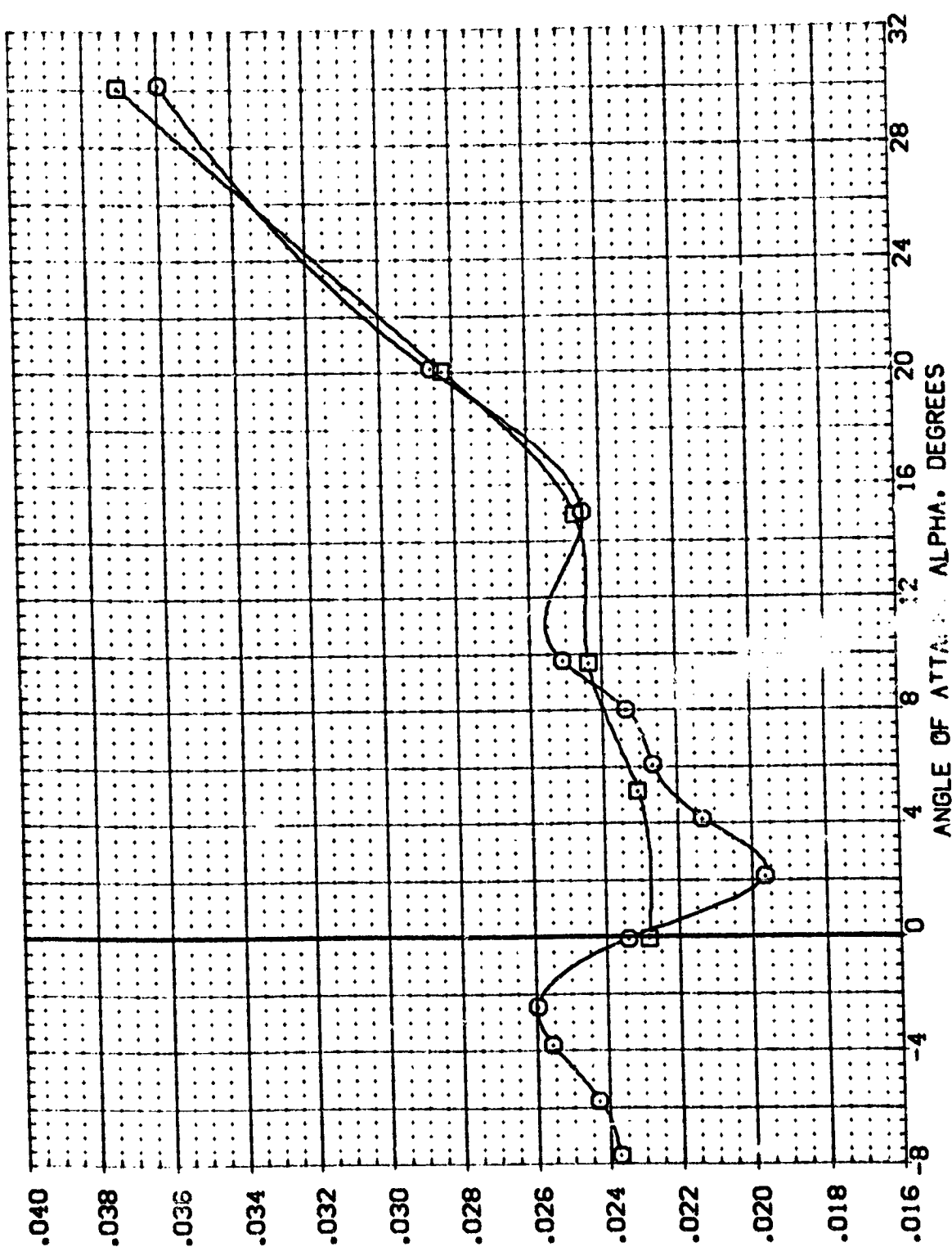


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(A)MACH = 5.26

DATA SET SYMBOL: (M97011) CONFIGURATION DESCRIPTION: CS T10 AT2 PLUVE OFF  
 (L97005) DATA NOT AVAILABLE  
 (M97004) AMES 3.5-169 IAI0  
 (V97005) AMES 3.5-169 IAI0

BETA: .000 AILRON: .000 ELEVON: .000 RUDDER: .000  
 .000 10.000 .000 .000  
 .000 .000 .000 .000  
 .000 10.000 .000 .000

REFERENCE INFORMATION:  
 SREF: 2650.0000 SQ.FT.  
 LREF: 1.250.0000 IN.  
 BREF: 976.6800 IN.  
 XWRP: 1073.4800 IN.  
 YWRP: .0000 IN.  
 ZWRP: 400.0000 IN.  
 SCALE: .0100

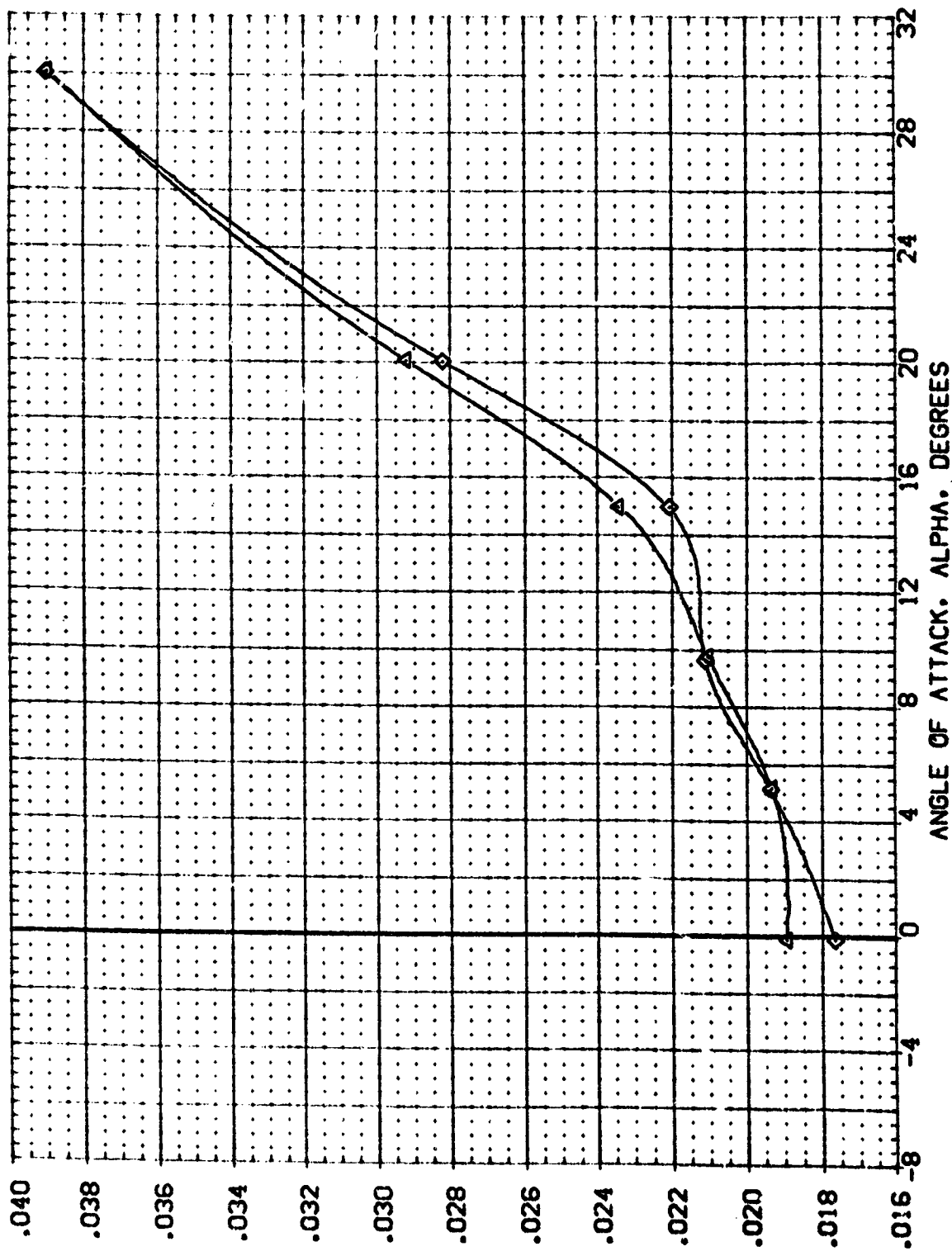


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(B)MACH = 7.32

DATA SET SYMBOL: (M87011), (U87005), (M87004), (U87003)  
 CONFIGURATION DESCRIPTION: ANES 3-5-169 IA10, ANES 3-5-169 IA10  
 DATA NOT AVAILABLE, DATA NOT AVAILABLE  
 BETA: .000, .000, .000, .000  
 AIRLON: .000, 10.000, 10.000  
 ELEVON: .000, .000, .000  
 FLUDDER: .000, .000, .000  
 REFERENCE INFORMATION: SREF 2690.0000, LREF 1290.0000, BREF 936.6800, XMRP 1076.4800, YMRP 400.0000, ZMRP 400.0000, SCALE .0100  
 SO. FT.: IN., IN., IN., IN., IN.

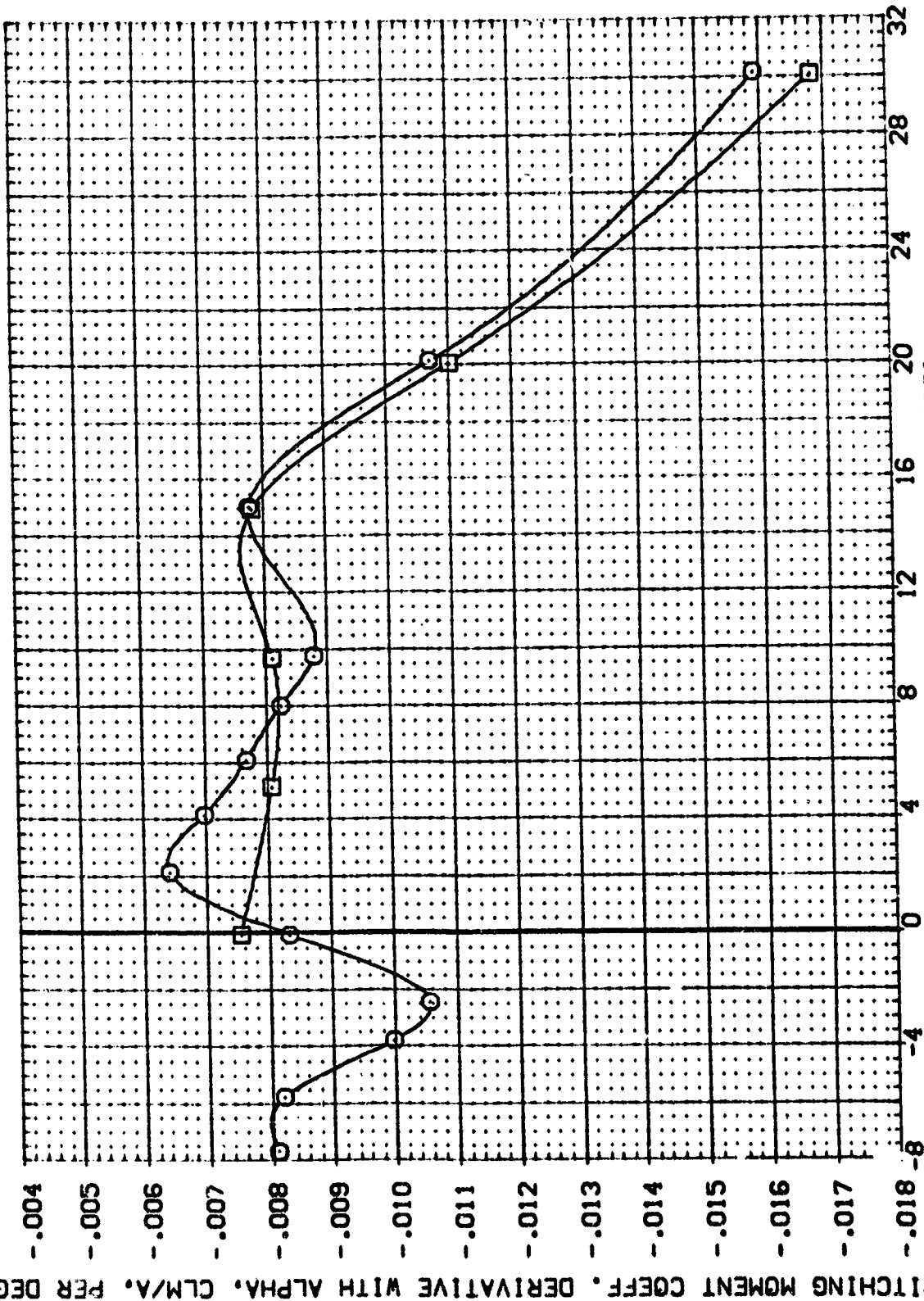


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(A)MACH = 5.26



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (M87011) DATA NOT AVAILABLE  
 (M87005) DATA NOT AVAILABLE  
 (M87004) APES 3.5-169 (A10 09 T10 AT2 PLUME OFF  
 (M87005) APES 3.5-163 (A10 09 T10 AT2 PLUME OFF

BETA AILRON ELEVON RUDDER REFERENCE INFORMATION  
 .000 .000 .000 SREF 2690.0000 SO.FT.  
 .000 .000 .000 LREF 1290.0000 IN.  
 .000 .000 .000 BREF 936.6800 IN.  
 .000 .000 .000 XMRP 1076.4800 IN.  
 .000 .000 .000 YMRP .0000 IN.  
 .000 .000 .000 ZMRP 400.0000 IN.  
 SCALE .0100

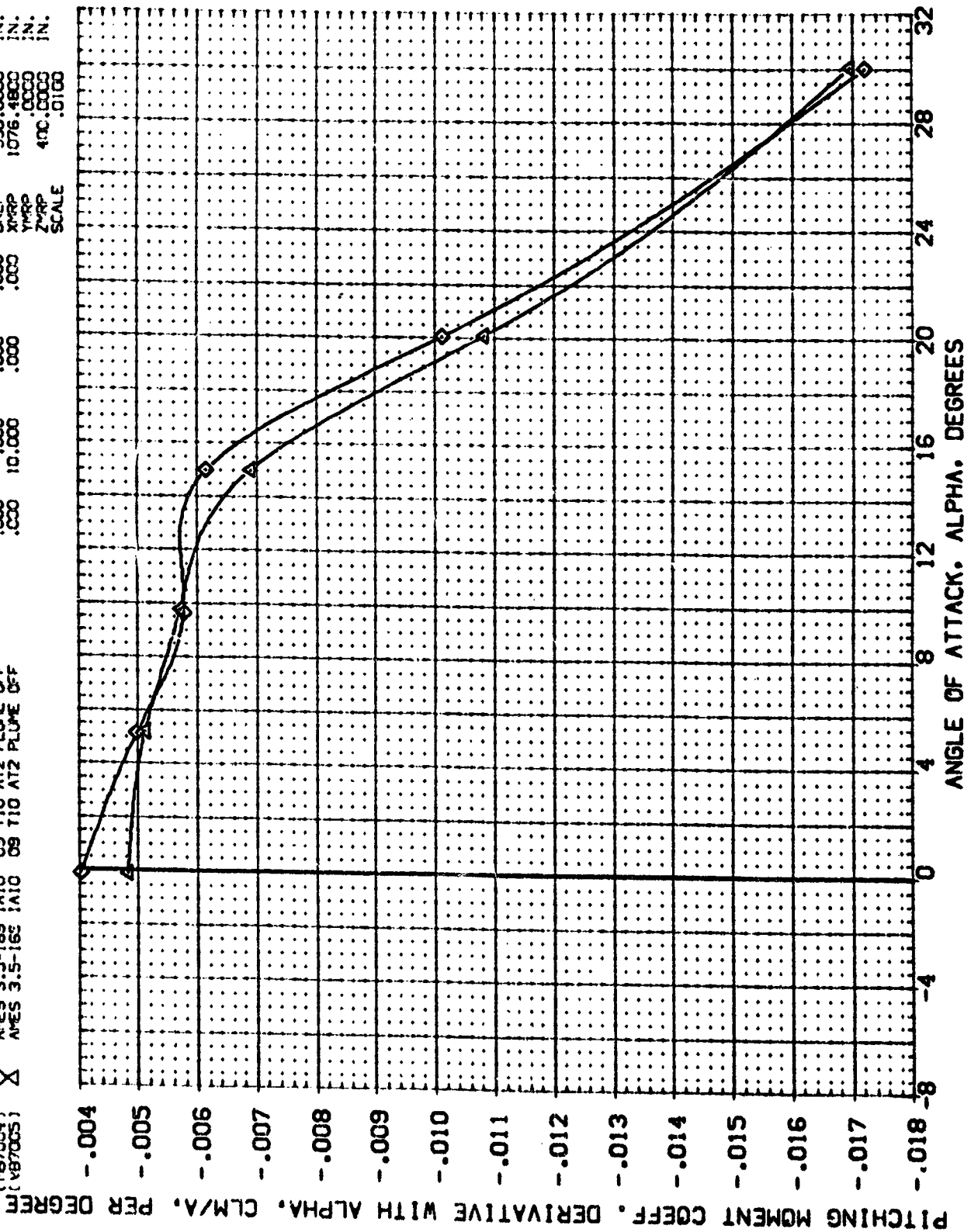


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(B)MACH = 7.32



DATA SET SYMBOL    CONFIGURATION DESCRIPTION    REFERENCE INFORMATION

(187011)	NES 3.5-189 IA10 OS T10 AT2 PLVE OFF	SREF 2680.0000 SO.FT.
(187005)	NES 3.5-189 IA10 OS T10 AT2 PLVE OFF	LREF 1290.0000 IN.
(187004)	DATA NOT AVAILABLE	BREF 936.6800 IN.
(187005)	DATA NOT AVAILABLE	XMRP 1076.4800 IN.
		YMRP 400.0000 IN.
		ZMRP 400.0000 IN.
		SCALE .0100

BETA    AILRON    ELEVON    RUDDER

.000	.000	.000	.000
.000	10.000	.000	.000
.000	.000	.000	.000
.000	10.000	.000	.000

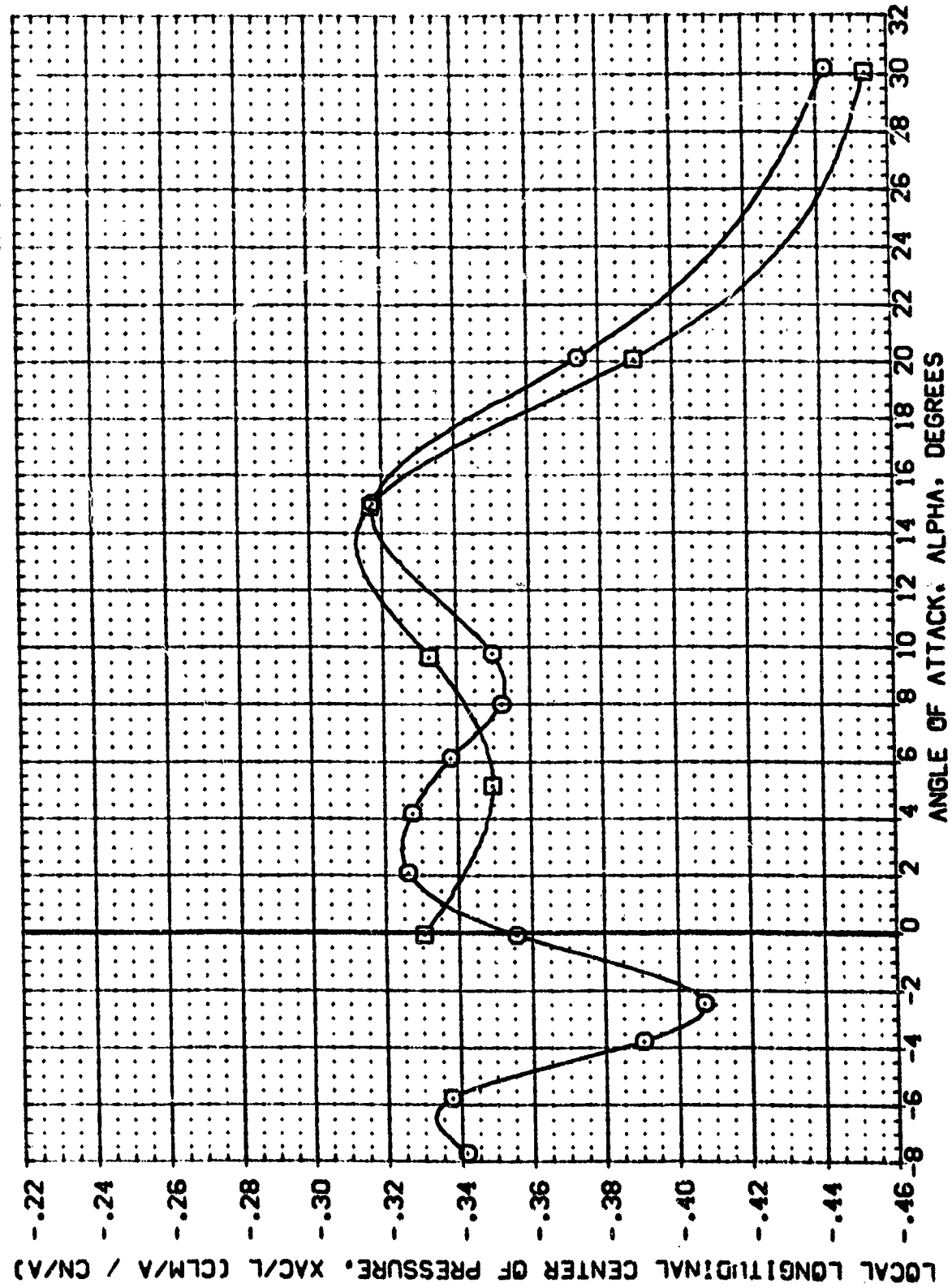


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(A)MACH = 5.26





AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF (087005)

SYMBOL MACH BETA PARAMETRIC VALUES

○ 5.300 .000

□ 7.300

REFERENCE INFORMATION

SREF 2680.0000 SO.FT.

LREF 1250.0000 IN.

BREF 936.6800 IN.

XMRP 1076.4800 IN.

YMRP .0000 IN.

ZMRP 400.0000 IN.

SCALE .0100

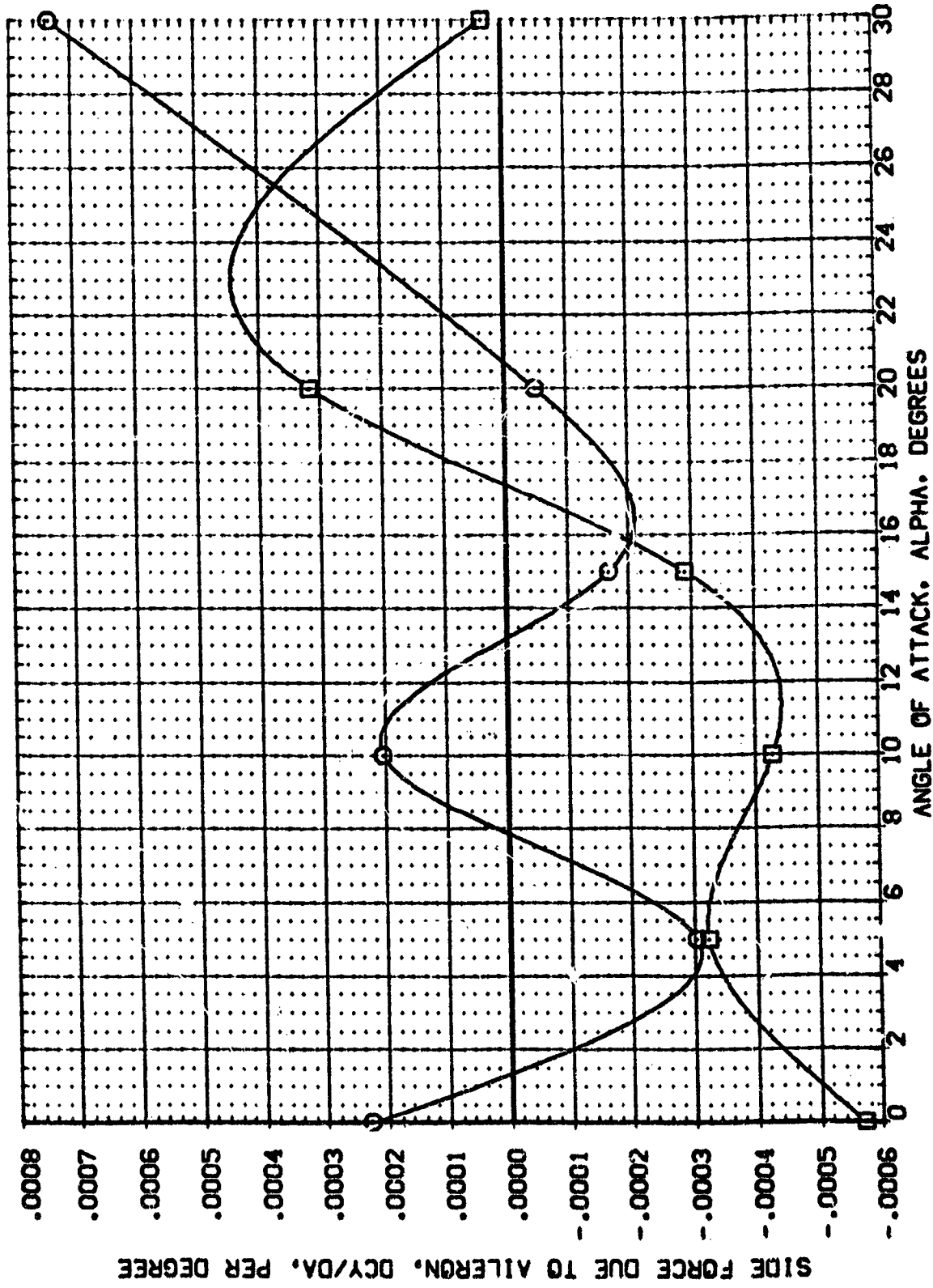


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

AMES 3.5-169 IA10 09 T10 AT2 PLUME OFF (QB7005)

SYMBOL MACH BETA PARAMETRIC VALUES  
 ○ 5.300 .000  
 □ 7.300

REFERENCE INFORMATION  
 SREF 2690.0000 SO.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 YMRP 1076.4800 IN.  
 ZMRP .0000 IN.  
 SCALE 400.0000 IN.

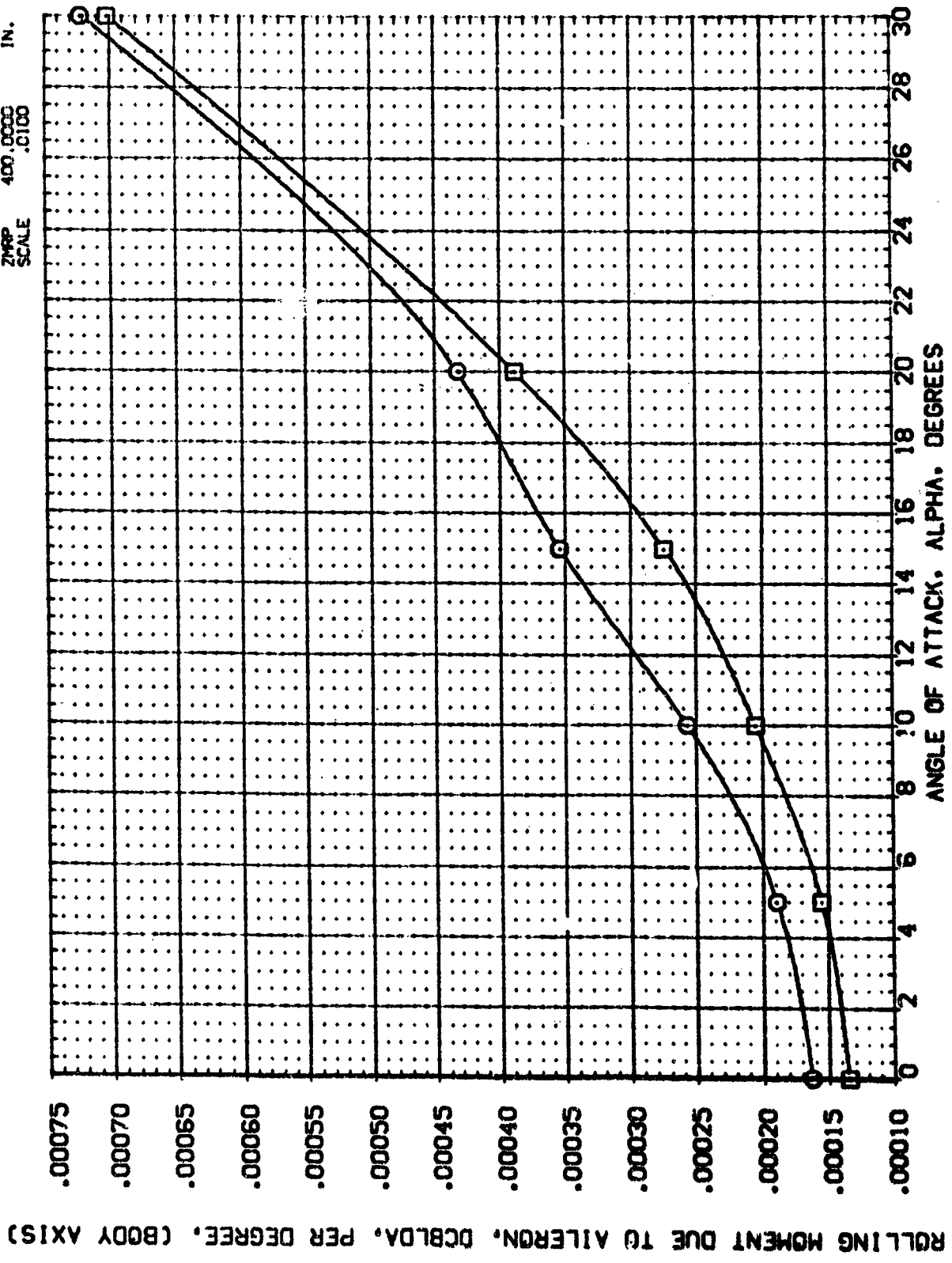


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.



AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF (087005)

SYMBOL    MACH    BETA    PARAMETRIC VALUES

□        5.300    .000

○        7.300

REFERENCE INFORMATION

SREF    2690.0000    SQ.FT.

LREF    1250.0000    IN.

BREF    936.8600    IN.

XMRP    1076.4800    IN.

YMRP    400.0000    IN.

ZMRP    400.0000    IN.

SCALE    .0100

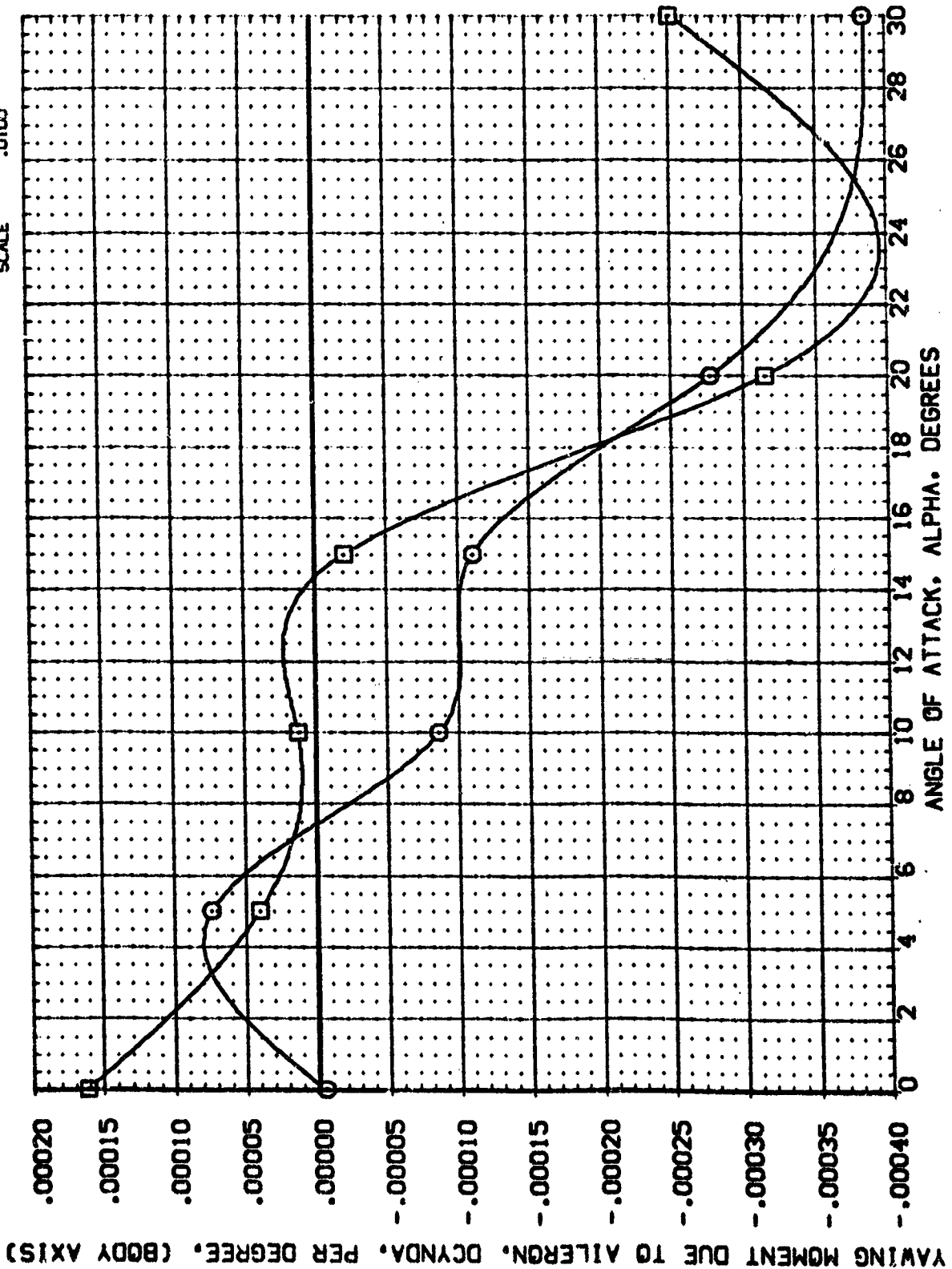


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6900 IN.  
 YWRP 1076.4900 IN.  
 ZWRP 400.0000 IN.  
 SCALE .0100

SYMBOL MACH BETA PARAMETRIC VALUES  
 ○ 5.300 .000  
 □ 7.300

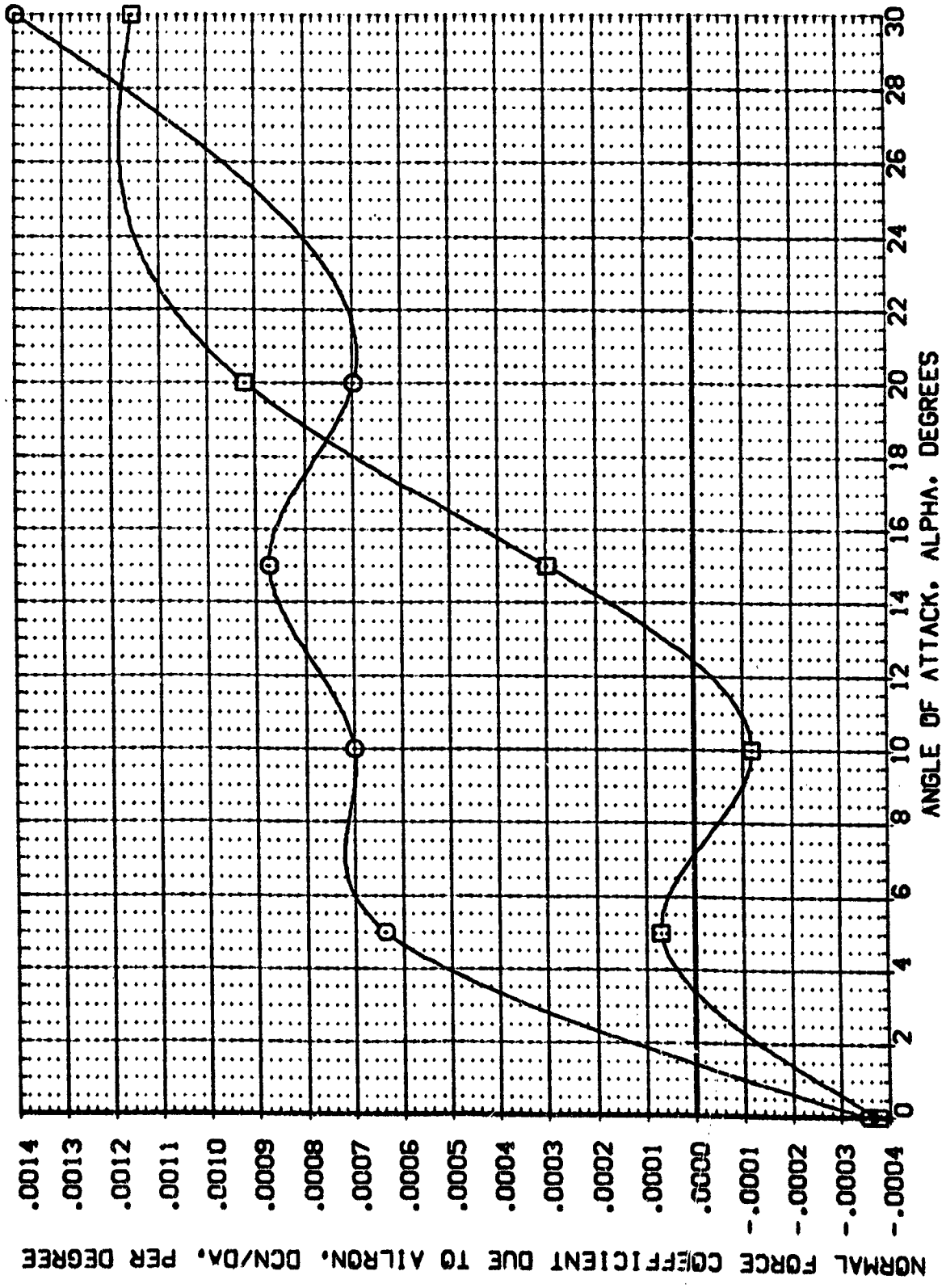


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(QB7005)

AMES 3.5-169 IA10 09 T10 AT2 PLUME OFF

SYMBOL MACH BETA PARAMETRIC VALUES  
O 5.300 .000  
□ 7.300

REFERENCE INFORMATION

SREF 2690.0000 SQ.FT.  
LREF 1290.0000 IN.  
BREF 936.6800 IN.  
XPRP 1076.4800 IN.  
YPRP .0000 IN.  
ZPRP 400.0000 IN.  
SCALE .0100

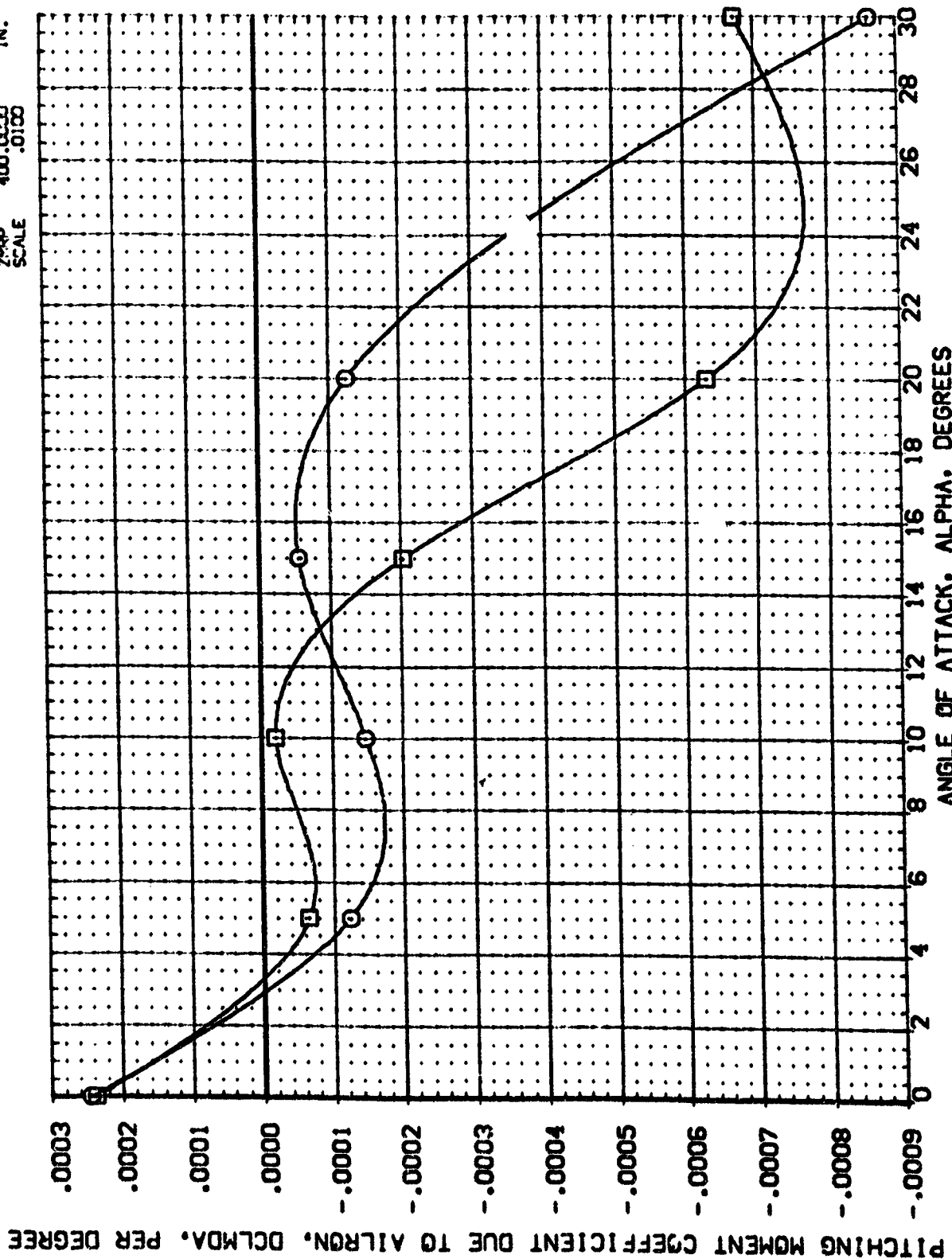


FIG 11 ELEVON EFFECTIVENESS SUMMARY.



SYMBOL MACH BETA PARAMETRIC VALUES  
 □ 5.300 .000  
 ○ 7.300

REFERENCE INFORMATION  
 SREF 2650.0000 SQ.Ft.  
 REF 1290.0000 IN.  
 REF 936.6800 IN.  
 YPRP 1076.4800 IN.  
 ZPRP 400.0000 IN.  
 SCALE 400.0100

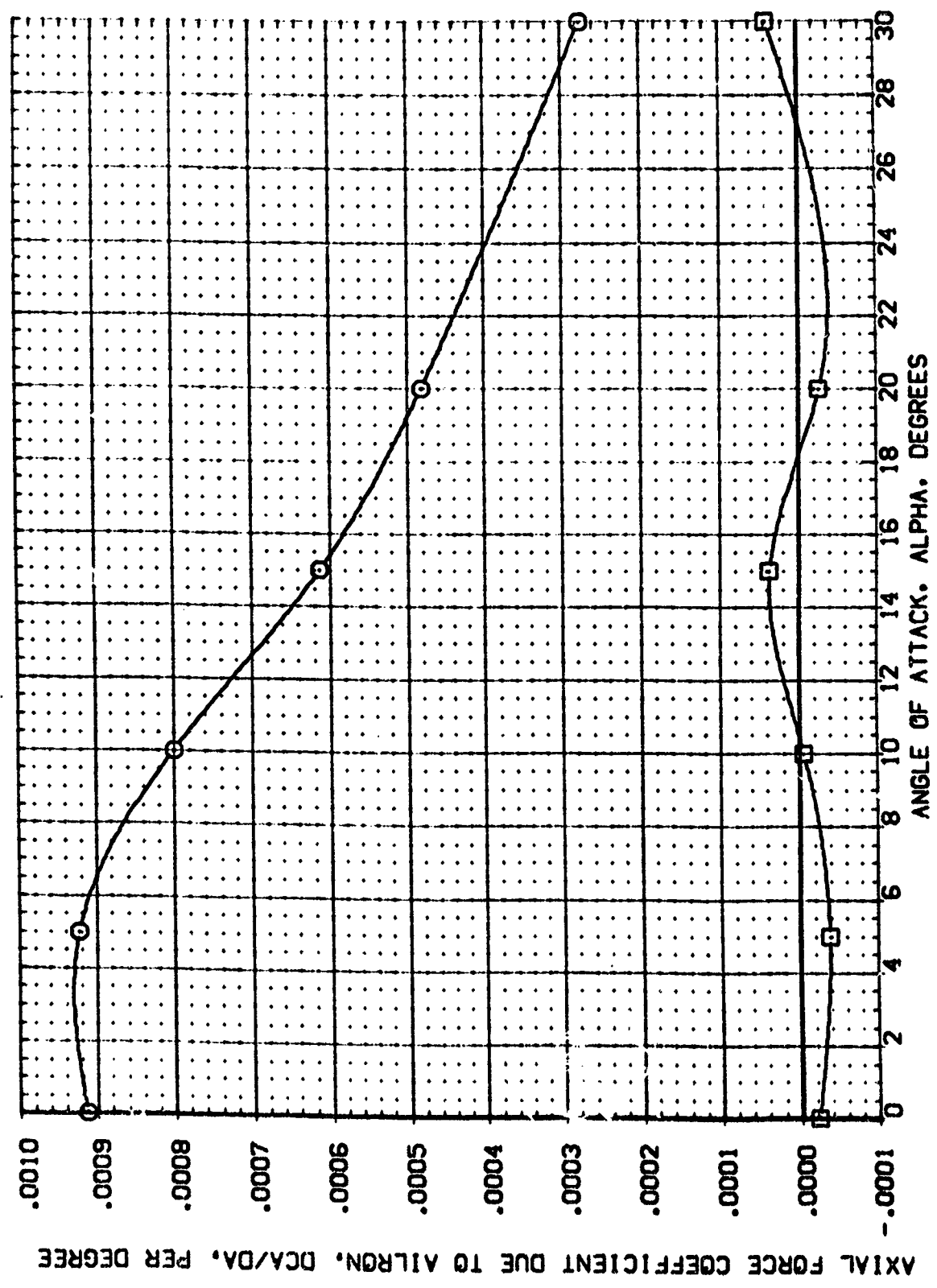


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.



(087002)

AMES 3.5-169 IA10 09 T10 AT2 PLUME OFF

SYMBOL MACH BETA

□ 5.300 5.000

○ 7.300

PARAMETRIC VALUES

REFERENCE INFORMATION

SREF 2690.0000 SQ.FT.

LREF 1290.0000 IN.

BREF 936.6800 IN.

XPRP 1076.4800 IN.

YPRP .0000 IN.

ZPRP 400.0000 IN.

SCALE .0100

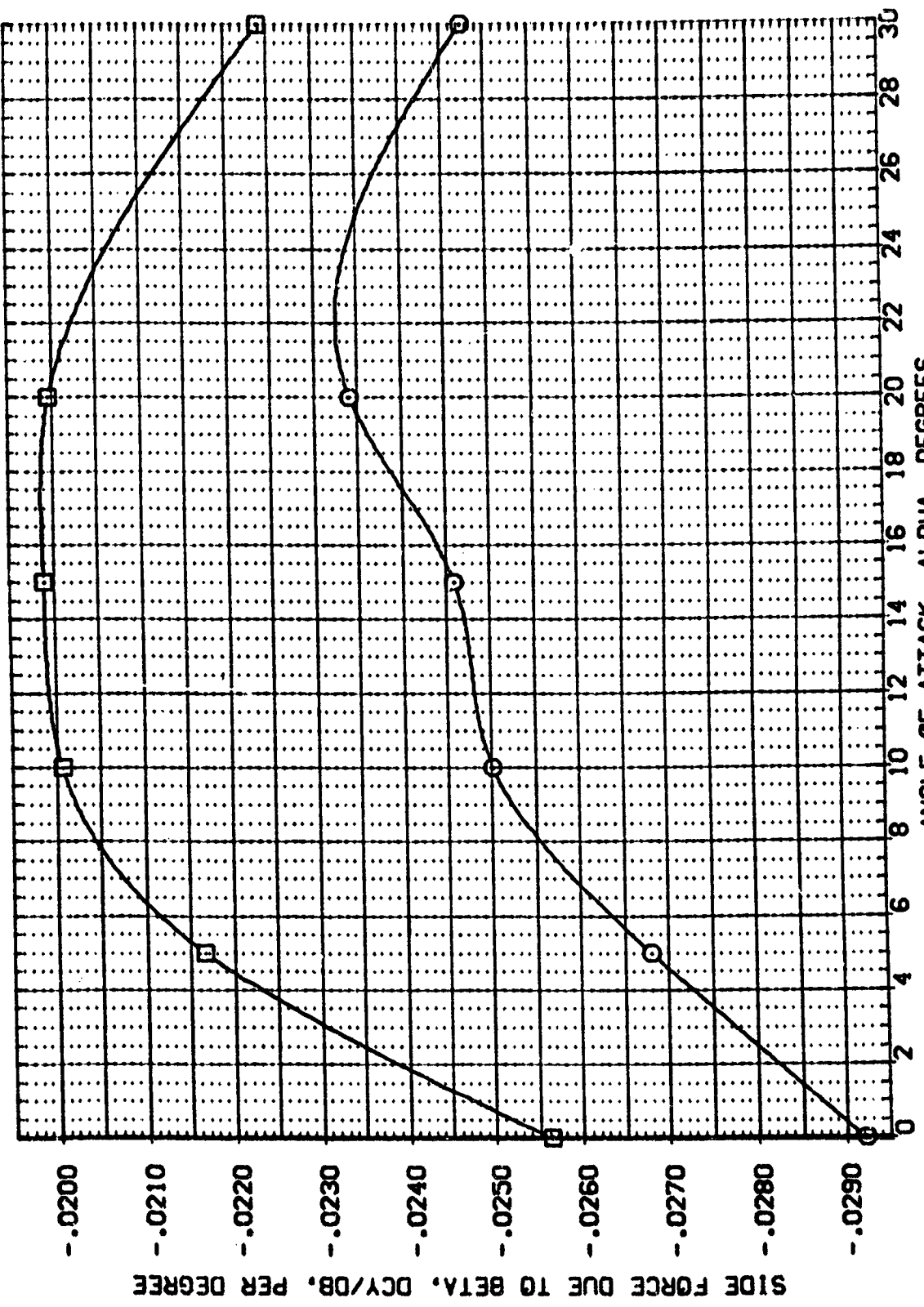


FIG. 12 SUMMARY OF ANGLE OF ATTACK EFFECTS ON LATERAL-DIRECTIONAL CHARACT.

SYMBOL MACH BETA  
 ○ 5.300  
 □ 7.300

PARAMETRIC VALUES  
 MACH 5.000

REFERENCE INFORMATION  
 SREF 2690.0000 SO.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

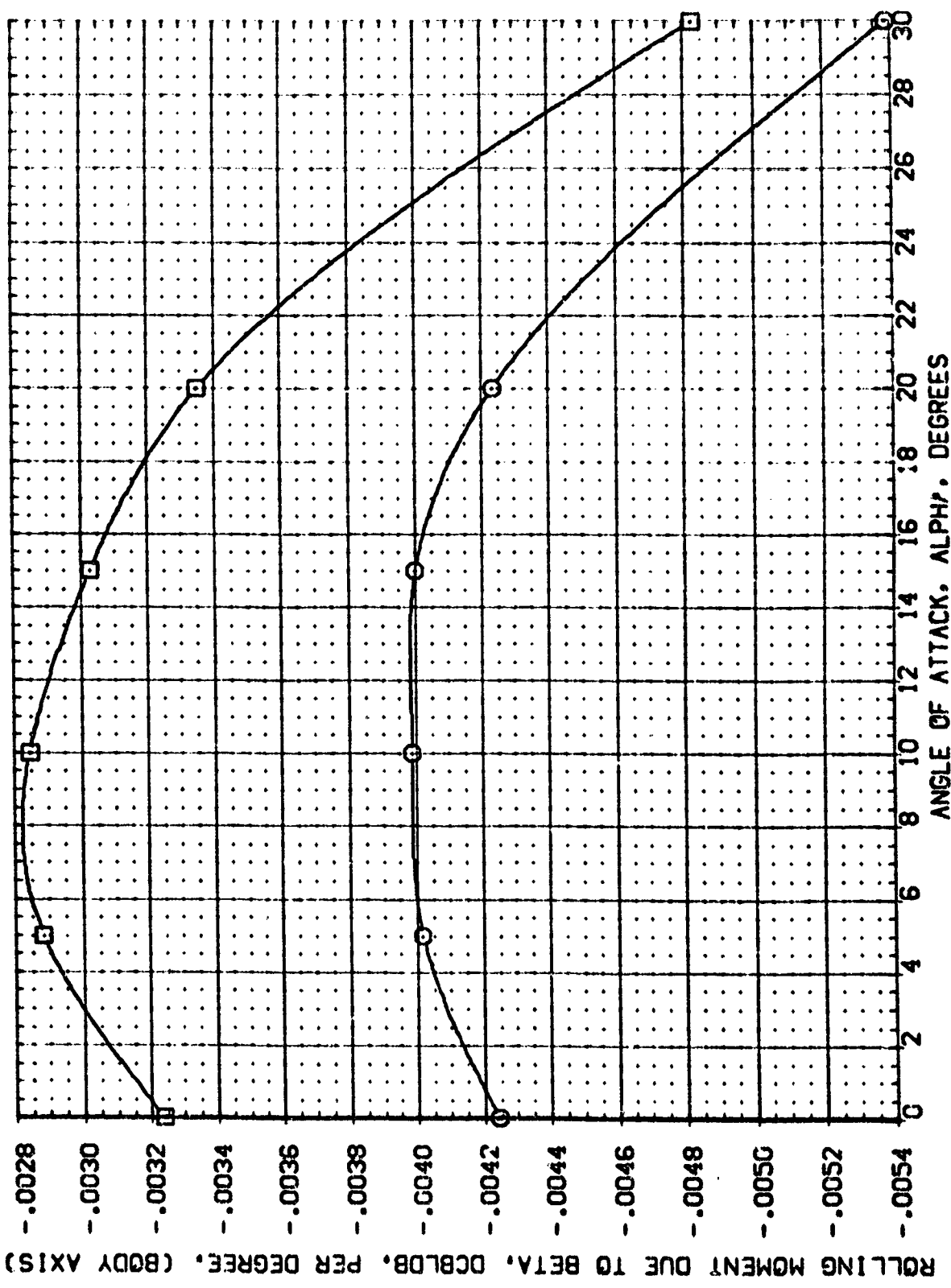


FIG. 12 SUMMARY OF ANGLE OF ATTACK EFFECTS ON LATERAL-DIRECTIONAL CHARACT.

AMES 3.5-169 IA10 09 T10 AT2 PLUME OFF (QB7002)

SYMBOL MACH BETA  
 ○ 5.300  
 □ 7.300

PARAMETRIC VALUES  
 5.000

REFERENCE INFORMATION  
 SREF 2690.0000 SC.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1576.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

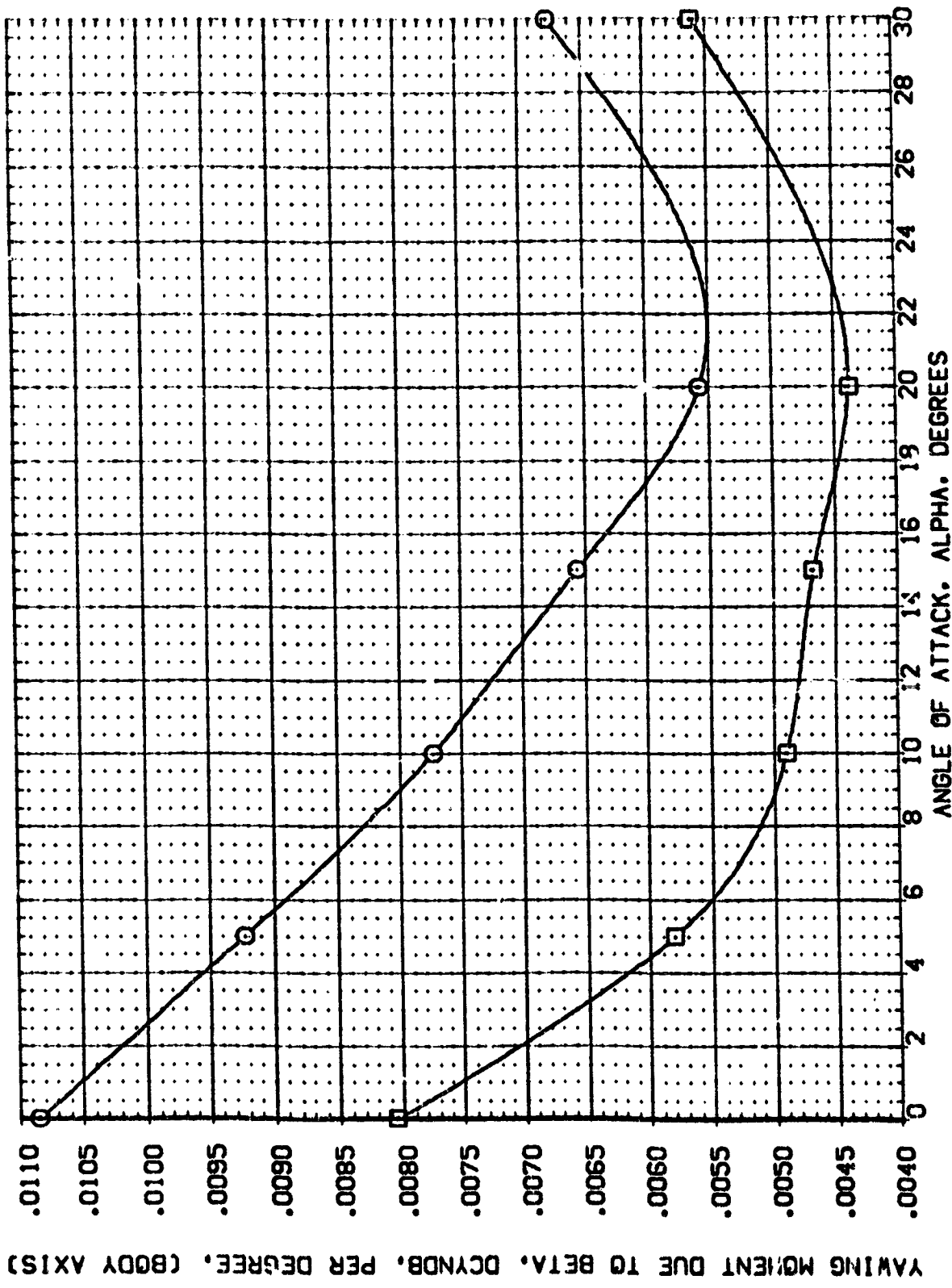


FIG. 12 SUMMARY OF ANGLE OF ATTACK EFFECTS ON LATERAL-DIRECTIONAL CHARACTER.

APPENDIX

TABULATED SOURCE DATA

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Tabulations of plotted data are available on request from  
Data Management Systems

DATE 05 DEC 75

0607021 ( 28 NOV 75 )

TABLATED SOURCE ARC 3.5-169, 1A10  
AVES 3.5-169 1A10 CS T10 ATR PLUKE OFF

PARAMETRIC DATA

BETA = .000 AILRON = .000  
ELEVON = .000 BOFLAP = .000  
SPDBRK = .000 RUDDER = .000

REFERENCE DATA

REF = 2680.0000 50.FT. WARP = 1076.4000 IN.  
LREF = 1290.0000 IN. WARP = .0000 IN.  
BREF = 936.6000 IN. WARP = 400.0000 IN.  
SCALE = .0100

RAVL = 4.13 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CN	CY	CA	CLM	CYN	CEL	CL	CD	L/D
7.320	-9.590	-23462	-01010	.21747	.09320	.03863	-.00193	-.21486	.25691	-.89671
	-7.387	-21875	-.00964	.20424	.06651	.03961	-.00201	-.19068	.25067	-.82661
	-5.356	-17338	-.00814	.19355	.07138	.02259	-.00162	-.15463	.20983	-.74046
	-3.384	-13376	-.00610	.18164	.05909	.02139	-.00149	-.12280	.18922	-.64899
	-1.516	-.09334	-.00727	.17063	.04450	.02151	-.00136	-.08079	.17306	-.51303
	.165	-.09664	-.00466	.16151	.03326	.02063	-.00106	-.05710	.16134	-.35393
	2.413	-.01616	-.00714	.14979	.02163	.02127	-.00107	-.02245	.14896	-.19070
	4.291	.01646	-.00950	.14169	.01446	.02066	-.00097	.00556	.14254	.03904
	6.507	.05678	-.00594	.13785	.02404	.02103	-.00075	.04560	.14339	.28432
	8.591	.09489	-.00510	.13027	-.00626	.02048	-.00057	.07417	.14295	.51680
	10.606	.13597	-.00444	.12494	-.01785	-.00020	-.00043	.11165	.14783	.74851
GRADIENT		.01934	.00006	-.00516	-.00573	-.00006	.00007	.01654	-.02600	.08932

MACH 7.320  
CAAFD .162389  
OMAFD -.060034  
CLMAFD .034261  
MAC/L -.296205

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR.

REFERENCE DATA

SREF = 2890.0000 SQ.FT. WARP = 1076.4600 IN.  
 LREF = 1290.0000 IN. WARP = .0000 IN.  
 SREF = 936.6600 IN. ZARP = 400.0000 IN.  
 SCALE = .0100

RUN NO. 0/ 0 RMV = 2.37 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CN	CY	CA	CLM	CYN	CEL	CL	CD	L/D
5.280	-1.110	-0.6013	-1.6291	.17932	.03937	.05904	-.02304	-.05979	.17643	-.30986
5.280	5.108	.06267	-1.4218	.15707	-.00997	.04669	-.02126	.04843	.16203	.27875
5.280	9.683	.16683	-1.3218	.14344	-.00969	.03851	-.02072	.14030	.16948	.77852
5.280	14.980	.29331	-1.2622	.12927	-.06380	.03216	-.02105	.24993	.20069	1.18672
5.280	20.084	.42996	-1.2280	.12131	-.13296	.02829	-.02236	.36216	.26159	1.33761
5.280	30.044	.76390	-1.1716	.11844	-.27161	.03140	-.02753	.60370	.48999	1.22156
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

RUN NO. 0/ 0 RMV = 4.98 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CN	CY	CA	CLM	CYN	CEL	CL	CD	L/D
7.320	-0.066	-0.09910	-1.5144	.17141	.03106	.04654	-.01978	-.05890	.7148	-.32076
7.320	5.188	.04669	-1.3109	.14567	.00198	.03472	-.01785	.03362	.14932	.20996
7.320	9.821	.13969	-1.2085	.12977	-.02374	.02831	-.01754	.11551	.15170	.71515
7.320	15.124	.25675	-1.1612	.11904	-.09804	.02442	-.01820	.21712	.18153	1.13876
7.320	20.180	.39229	-1.1009	.11482	-.10433	.02444	-.02084	.32871	.24282	1.30565
7.320	30.025	.74373	-1.0207	.11411	-.24890	.03353	-.02880	.56654	.47092	1.22372
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

MACH 5.280 CLMWF0 .058434  
 7.320 .171065 .050689  
 7.320 .171065 -.057756

PARAMETRIC DATA

BETA = 5.000 AILRON = .000  
 ELEVON = .000 SOFLAP = .000  
 SPDRK = .000 RUDDER = .000

ARC 3.5-109 1A10 ON T10 ATE PLANE OFF

097002) ( 28 NOV 73 )

REFERENCE DATA

SKEW = 2000.0000 90. PT.    100P = 1076.4000 IN.  
 LEW = 2000.0000 IN.        100P = .0000 IN.  
 SPK = 906.0000 IN.        200P = 400.0000 IN.  
 SCALE = .0100

PARAMETRIC DATA

BETA = 5.000    AIRLON = .000  
 ELEVON = .000    BOFLAP = .000  
 SPDRK = .000    RUDDER = .000

RUN NO. 0/ 0 RVAL = 2.37 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CLM/A	CM/A	XAC/L
5.200	-1.110	-.00648	.02382	-.39606
5.200	5.108	-.00796	.02297	-.34651
5.200	9.093	-.00792	.02298	-.34490
5.200	14.960	-.00872	.02495	-.34839
5.200	20.064	-.01115	.02906	-.37865
5.200	30.044	-.01531	.03591	-.42817
	GRADIENT	.00000	.00000	.00000

RUN NO. 0/ 0 RVAL = 4.99 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CLM/A	CM/A	XAC/L
7.320	-.086	-.00564	.02036	-.27689
7.320	5.168	-.00532	.01901	-.26036
7.320	9.621	-.00586	.02090	-.28015
7.320	15.024	-.00754	.02392	-.31540
7.320	20.180	-.01109	.02979	-.37233
7.320	30.023	-.01642	.03866	-.42463
	GRADIENT	.00000	.00000	.00000



DATE 05 DEC 75

TABULATED SOURCE ARC 3.5-169 IA1D

(887003) ( 21 SEP 75 )

REFERENCE DATA

WARP = 2880.0000 90.FT. WARP = 1076.4800 IN.  
LWAP = 1290.0000 IN. WARP = .0000 IN.  
SWAP = 936.6800 IN. ZWAP = 400.0000 IN.  
SCALE = .0100

PARAMETRIC DATA

BETA = 5.000 AILRON = .000  
ELEVON = .000 BOPLAP = .000  
SPUWER = .000 RUDDER = .000

RM/L = 3.79 GRADIENT INTERVAL = -5.00/ 5.00

MAON	7.320	ALPHA	CN	CV	CA	CLM	CYN	CEL	CL	CD	L/D
GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
30.074	.72961	.11784	.11803	.10127	.02464	.02951	.02904	.46571	1.20847	1.24389	.00000
20.136	.39626	.11784	.11803	.10127	.02464	.02951	.02904	.46571	1.20847	1.24389	.00000
14.955	.23489	.11784	.11803	.10127	.02464	.02951	.02904	.46571	1.20847	1.24389	.00000
9.726	.13572	.11784	.11803	.10127	.02464	.02951	.02904	.46571	1.20847	1.24389	.00000
5.161	.04615	.11784	.11803	.10127	.02464	.02951	.02904	.46571	1.20847	1.24389	.00000
-.123	-.05986	.11784	.11803	.10127	.02464	.02951	.02904	.46571	1.20847	1.24389	.00000
ALPHA	7.320	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
CV	-.15166	.17526	.03197	.04709	-.01978	-.02949	.17539	-.31983	.17539	-.31983	.17539
CA	.14732	.13136	-.02992	.02813	-.01696	.11157	.15241	.69902	.15241	.69902	.15241
CLM	.03197	.03473	.02813	.02464	.02951	.02904	.46571	1.20847	1.24389	.00000	.00000
CYN	.04709	.02813	.02464	.02951	.02904	.46571	1.20847	1.24389	.00000	.00000	.00000
CEL	-.01978	-.01696	.11157	.15241	.69902	.15241	.69902	.15241	.69902	.15241	.69902
CL	.03271	.21534	.32273	.57343	.00000	.00000	.00000	.00000	.00000	.00000	.00000
CD	.18087	.24191	.46571	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
L/D	.20253	1.13003	1.20847	1.24389	.00000	.00000	.00000	.00000	.00000	.00000	.00000

(887004) ( 26 NOV 75 )

AVES 3.5-169 IA1D CB T10 AT2 FLUME OFF

REFERENCE DATA

WARP = 2880.0000 90.FT. WARP = 1076.4800 IN.  
LWAP = 1290.0000 IN. WARP = .0000 IN.  
SWAP = 936.6800 IN. ZWAP = 400.0000 IN.  
SCALE = .0100

PARAMETRIC DATA

BETA = .000 AILRON = .000  
ELEVON = .000 BOPLAP = .000  
SPUWER = .000 RUDDER = .000

RM/L = 4.76 GRADIENT INTERVAL = -5.00/ 5.00

MAON	7.320	ALPHA	CN	CV	CA	CLM	CYN	CEL	CL	CD	L/D
GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
30.060	.73487	.11291	.11291	-.25722	.00446	-.00484	-.00484	.57030	.46572	.00000	.00000
20.054	.37997	.11291	.11291	-.25722	.00446	-.00484	-.00484	.57030	.46572	.00000	.00000
14.967	.23819	.11291	.11291	-.25722	.00446	-.00484	-.00484	.57030	.46572	.00000	.00000
9.661	.14177	.11291	.11291	-.25722	.00446	-.00484	-.00484	.57030	.46572	.00000	.00000
5.165	.04696	.11291	.11291	-.25722	.00446	-.00484	-.00484	.57030	.46572	.00000	.00000
-.112	-.04623	.11291	.11291	-.25722	.00446	-.00484	-.00484	.57030	.46572	.00000	.00000
ALPHA	7.320	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
CV	-.02252	.16104	.02995	.00680	-.00354	-.00354	.16113	-.26510	.16113	-.26510	.16113
CA	.14100	.14100	.00601	.00596	-.00323	.11654	.14491	.29505	.14491	.29505	.14491
CLM	.02995	.00601	.00596	.00398	-.00278	.21743	.17873	1.21639	.17873	1.21639	.17873
CYN	.00680	.00596	.00398	.00129	-.00272	.31816	.29682	1.34316	.29682	1.34316	.29682
CEL	-.00354	-.00323	.11654	.14491	.29505	.14491	.29505	.14491	.29505	.14491	.29505
CL	.02813	.02464	.02951	.02904	.46571	.00000	.00000	.00000	.00000	.00000	.00000
CD	.18087	.24191	.46571	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
L/D	.20253	1.13003	1.20847	1.24389	.00000	.00000	.00000	.00000	.00000	.00000	.00000

MAON 7.320 CAWFO .160611 CWFWD .028436 CLMWF0 .028436

TABLATED SOURCE ARC 3.5-169, 1A10

(997004) ( 28 NOV 75 )

AXES 3.5-169 1A10 OR T10 ATE PLANE OF

REFERENCE DATA

XREF = 2080.0000 90. FT.    XREF = 1076.4000 IN.  
 YREF = 1250.0000 IN.    YREF = .0000 IN.  
 ZREF = 506.0000 IN.    ZREF = 400.0000 IN.  
 SCALE = .0100

PARAMETRIC DATA

BETA = .000    AILRON = .000  
 ELEVON = .000    BOFLAP = .000  
 SPDRK = .000    RUDDER = .000

RM/L = 4.78    GRADIENT INTERVAL = -5.00/ 9.00

MACH = 7.350

ALPHA	CLM/A	CMVA	XAC/L
-.112	-.00403	.01764	-.22933
5.165	-.00498	.01937	-.23715
9.661	-.00577	.02113	-.27306
14.967	-.00614	.02209	-.27769
20.054	-.01012	.02621	-.33679
30.080	-.01721	.03907	-.44065
GRADIENT	.00000	.00000	.00000

DATE 05 DEC 75

TABULATED SOURCE ARC 3.5-169, 1A10

ANES 3.5-169 1A10 CO T10 ATI PLUME OFF

(NS7005) ( 28 NOV 75 )

REFERENCE DATA

SREF = 2880.0000 30. FT. XAPP = 1076.4800 IN.  
 LREF = 1280.0000 IN. YAPP = .0000 IN.  
 BREF = 936.6800 IN. ZAPP = 400.0000 IN.  
 SCALE = .0100

BETA = .000 AILRON = 10.000  
 ELEVON = .000 BOFLAP = .000  
 SPOBRK = .000 RUDDER = .000

PARAMETRIC DATA

GRADIENT INTERVAL = -5.00/ 5.00

WACH	ALPHA	CN	CY	CA	CLM	CYN	CEB	CL	CD	L/D
5.280	-0.075	-0.0148	-0.01354	.1774	.09867	.00417	-.00006	-.05125	.17781	-.28821
5.280	5.130	.06776	-0.01084	.16366	-.00348	.00109	.00083	.05286	.16956	.31264
5.280	9.667	.17475	-0.04039	.14756	-.04096	-.00166	.00184	.14747	.17481	.84360
5.280	14.934	.30368	-0.04443	.13933	-.08218	-.00196	.00259	.26068	.20958	1.24380
5.280	20.081	.45758	-0.05579	.12965	-.12827	-.00247	.00327	.36782	.26824	1.57124
5.280	30.068	.77857	-0.04434	.11607	-.27694	-.00691	.00672	.61564	.49054	1.25303
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

GRADIENT INTERVAL = -5.00/ 5.00

WACH	ALPHA	CN	CY	CA	CLM	CYN	CEB	CL	CD	L/D
7.320	-0.080	-0.04881	-0.02885	.16079	.03127	.00819	-.00221	-.04955	.18767	-.30858
7.320	5.200	.05105	-0.02458	.14227	.00323	.00659	-.00169	.05812	.14432	.26417
7.320	9.771	.14258	-0.02120	.12382	-.01941	.00368	-.00078	.11916	.14819	.80410
7.320	15.011	.25970	-0.01855	.11865	-.03202	.00110	-.00001	.22182	.17994	1.22611
7.320	20.115	.39096	-0.01480	.11265	-.09531	-.00042	.00020	.32799	.24010	1.36658
7.320	30.100	.74748	-0.01681	.11329	-.24431	.00207	.00217	.59987	.47289	1.24758
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

WACH CAAPP CAPP CAPP  
 5.280 .17758 .16320 .158104  
 7.320 .16320 .148045 .090820

DATE 03 DEC 73

TABULATED SOURCE ARC 3.5-100. 1A1D

(P87005) ( 28 NOV 73 )

ARC 3.5-100 1A1D C9 T1D AT2 PLUME OFF

REFERENCE DATA

REF = 2000.0000 50.FT.    WHP = 1076.4800 IN.  
 LREF = 1290.0000 IN.    WHP = .0000 IN.  
 SREF = 936.6000 IN.    ZHP = 400.0000 IN.  
 SCALE = .0100

PARAMETRIC DATA

BETA = .000    ATLRON = 10.000  
 ELEVON = .000    BOFLAP = .000  
 SPOBRK = .000    RUDDER = .000

RUN NO. 0/ 0 RW/L = 1.92 GRADIENT INTERVAL = -5.00/ 5.00

WACH	ALPHA	CL/WA	OW/A	YAC/L
5.260	-.075	-.00754	.02265	-.33016
5.260	5.130	-.00807	.02307	-.34660
5.260	9.667	-.00811	.02437	-.33260
5.260	14.934	-.00782	.02489	-.31675
5.260	20.081	-.01102	.02829	-.36942
5.260	30.068	-.01682	.03707	-.45372
	GRADIENT	.00000	.00000	.00000

RUN NO. 0/ 0 RW/L = 5.78 GRADIENT INTERVAL = -5.00/ 5.00

WACH	ALPHA	CL/WA	OW/A	YAC/L
7.320	-.069	-.00482	.01691	-.28491
7.320	5.200	-.00511	.01933	-.26449
7.320	9.771	-.00570	.02104	-.27066
7.320	15.011	-.00680	.02343	-.29454
7.320	20.115	-.01084	.02924	-.37069
7.320	30.101	-.01696	.03900	-.43496
	GRADIENT	.00000	.00000	.00000

REFERENCE DATA

REF = 2000.0000 SQ.FT. WARP = 1070.4000 IN.  
 LREF = 1250.0000 IN. WARP = .0000 IN.  
 BRP = 536.0000 IN. ZARP = 400.0000 IN.  
 SCALE = .0100

PARAMETRIC DATA

ALPHA = .000 AILLON = .000  
 ELEVON = .000 BOFLAP = .000  
 SPOBRK = .000 RUDDER = .000

MACH	BETA	ON	CY	CA	CLM	CYN	CEL	CL	CD	L/D
5.200	-9.500	-0.0015	.27782	.19193	.03200	-0.0061	.05023	-0.0015	.19193	-2.4000
5.200	-7.450	-0.0176	.20729	.18796	.03651	-0.07461	.02826	-0.0176	.18796	-2.0000
5.200	-5.475	-0.0449	.14449	.18393	.03944	-0.02778	.01923	-0.0449	.18393	-3.0000
5.200	-3.470	-0.0857	.07971	.17943	.04115	-0.02991	.01063	-0.0857	.17943	-3.5649
5.200	-1.600	-0.0011	.02447	.17706	.04246	-0.01167	.00354	-0.0011	.17706	-3.0000
5.200	.161	-0.0072	-.00391	.17646	.04163	-0.00334	-0.0722	-0.0072	.17646	-3.0000
5.200	2.272	-0.0406	-.00589	.16966	.04134	-0.02963	-0.0406	-0.0406	.16966	-3.4000
5.200	4.253	-0.0615	-.0217	.16032	.03968	-0.02338	-0.01944	-0.0615	.16032	-3.1146
5.200	6.343	-0.0577	-.01444	.15115	.03570	-0.02970	-0.02970	-0.0577	.15115	-2.0174
5.200	8.446	-0.0507	-.00402	.14392	.03363	-0.02976	-0.02976	-0.0507	.14392	-2.0192
5.200	10.502	-0.0367	-.00947	.12406	.03123	-0.02406	-0.02406	-0.0367	.12406	-2.0110
5.200	GRADIENT	.00061	.02990	.00110	-.00030	.01064	-.00396	.01061	.00110	.00674

GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CY	CA	CLM	CYN	CEL	CL	CD	L/D
7.300	-9.505	-0.04041	.28068	.16225	.02264	-0.06300	.03391	-0.04041	.16225	-2.2343
7.300	-7.532	-0.05042	.19350	.17431	.02715	-0.06260	.02321	-0.05042	.17431	-2.5510
7.300	-5.603	-0.05227	.13624	.16976	.02984	-0.04297	.01706	-0.05227	.16976	-2.6037
7.300	-3.402	-0.03291	.06432	.16467	.02926	-0.02615	.00992	-0.03291	.16467	-3.1222
7.300	-1.800	-0.05777	.03428	.16108	.03203	-0.01120	.00599	-0.05777	.16108	-3.5660
7.300	.152	-0.05882	-.00063	.16103	.03226	-0.01016	-0.01122	-0.05882	.16103	-3.0021
7.300	2.206	-0.05409	-.06309	.16295	.03031	-0.01716	-0.00763	-0.05409	.16295	-3.2715
7.300	4.248	-0.05087	-.11638	.16930	.02854	-0.04412	-0.01444	-0.05087	.16930	-2.6831
7.300	6.347	-0.03268	-.17104	.17395	.02632	-0.02332	-0.02213	-0.03268	.17395	-2.7457
7.300	8.405	-0.04064	-.22353	.17923	.02611	-0.02978	-0.02978	-0.04064	.17923	-2.3195
7.300	10.503	-0.03039	-.28432	.16895	.02313	-0.02128	-0.02900	-0.03039	.16895	-2.1363
7.300	GRADIENT	.00143	.02574	.00049	-.00018	.02769	-.00311	.00143	.00149	.00427

MACH YAC/L  
 5.200 17.457196  
 7.300 17.621084

DATE 03 DEC 73

04870071 ( 28 NOV 73 )

TABULATED SOURCE ARC 3.5-16J, 1A1D

ARC 3.5-169 1A1D C8 T1D AT2 PLUME OFF

PARAMETRIC DATA

ALPHA = .000 AILRON = .000  
ELEVON = .000 BOFLAP = .000  
SPYBRK = .000 RUDDER = 10.000

REFERENCE DATA

SRP = 2800.0000 98-FT. WARP = 1076.4000 IN.  
LWRP = 1290.0000 IN. VWRP = .0000 IN.  
SWRP = 936.0000 IN. ZWRP = 400.0000 IN.  
SCALE = .0100

MACH	BETA	ON	CY	CA	CLM	CYN	CEL	CL	CD	L/D
5.260	-9.561	.28229	.16631	.03378	-.10661	.04007	-.09899	.16631	-.25140	
5.260	-7.472	.20961	.18279	.03994	-.07904	.02938	-.05989	.18279	-.28722	
5.260	-5.426	.14580	.17957	.03916	-.05644	.02007	-.06377	.17957	-.33121	
5.260	-3.487	.08021	.17450	.04102	-.03318	.01183	-.06545	.17450	-.36854	
5.260	-1.526	.02840	.17066	.04177	-.01509	.00460	-.06561	.17066	-.39361	
5.260	1.66	-.01997	.16823	.04190	.00270	-.02212	-.06759	.16823	-.39879	
5.260	2.251	-.06203	.17421	.03980	.02492	-.01018	-.06342	.17421	-.35771	
5.260	4.266	-.06157	.17688	.03851	.04670	-.01600	-.06137	.17688	-.32930	
5.260	6.453	-.20335	.18209	.03502	.06911	-.02721	-.05694	.18209	-.27940	
5.260	8.454	-.28253	.18645	.03242	.08275	-.03613	-.05453	.18645	-.24490	
5.260	10.497	-.32705	.19250	.03054	.11791	-.04575	-.05381	.19250	-.21629	
GRADIENT		.00054	-.02883	.00044	-.00037	-.00386	.00154	.00044	-.00527	

MACH	BETA	ON	CY	CA	CLM	CYN	CEL	CL	CD	L/D
7.320	-9.595	-.05069	.26056	.17937	.02710	-.08798	.03549	-.09769	.17937	-.22962
7.320	-7.596	-.05413	.20056	.17374	.02927	-.06598	.02675	-.05413	.17374	-.27242
7.320	-5.481	-.05496	.14051	.16809	.03069	-.04575	.01817	-.03496	.16809	-.30415
7.320	-3.504	-.05712	.08468	.16353	.03180	-.02737	.01074	-.05712	.16353	-.33920
7.320	-1.656	-.06059	.03355	.16020	.03337	-.01225	.00474	-.06199	.16020	-.37599
7.320	1.110	-.03965	.110	.15951	.03311	-.00038	-.00037	-.03865	.15951	-.36768
7.320	2.268	-.05848	-.06251	.16233	.03145	.01666	-.00723	-.03648	.16233	-.34300
7.320	4.266	-.05394	-.11280	.16637	.03034	.03504	-.01374	-.05394	.16637	-.30947
7.320	6.367	-.05142	-.16579	.17184	.02830	.08032	-.02134	-.05142	.17184	-.27168
7.320	8.413	-.05180	-.22132	.17650	.02743	.08648	-.02686	-.05180	.17650	-.24645
7.320	10.495	-.05125	-.27949	.18392	.02638	.08941	-.03707	-.05125	.18392	-.21929
GRADIENT		.00055	-.02832	.00043	-.00027	-.00313	.00043	-.00495	.00043	

MACH YAC/L  
5.260 19.216957  
7.320 13.944568

AVES 3.5-169 IA1D C9 T10 ATZ PLUME ON

(RB7006) ( 21 SEP 73 )

REFERENCE DATA

SREF = 2660.0000 99. FT.    WARP = 1076.4000 IN.  
 LREF = 1250.0000 IN.    YWRP = .0000 IN.  
 BREF = 996.0000 IN.    ZWRP = 400.0000 IN.  
 SCALE = .0100

PARAMETRIC DATA

ALPHA = .000    AILCON = .070  
 ELEVON = .000    ROPFLAP = .000  
 SPOBRK = .000    RUDDER = 10.000

RUN NO. 0/ 0 RWAL = 2.05 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CT	CA	CLM	CYM	CSL	CL	CD	L/D
5.260	-0.2674	.26042	.17423	.03304	-.09826	.03687	-.06274	.17423	.17423	-.28712
5.260	-0.390	.20395	.17045	.03736	-.07012	.02604	-.06682	.17045	.17045	-.34174
5.260	-0.498	.13920	.16673	.03636	-.04639	.01677	-.06763	.16673	.16673	-.37826
5.260	-0.5312	.07194	.16172	.03682	-.02206	.00723	-.06723	.16172	.16172	-.40557
5.260	-0.582	.01131	.15801	.03611	.00003	-.00146	-.06643	.15801	.15801	-.41979
5.260	-0.6542	-.03457	.15752	.03689	.01320	-.00746	-.06542	.15702	.15702	-.41638
5.260	-0.6804	-.06331	.16236	.03780	.02427	-.00996	-.06504	.16236	.16236	-.39287
5.260	-0.7236	-.13564	.16877	.03753	.03927	-.01486	-.06236	.16877	.16877	-.34961
5.260	-0.76236	-.20393	.17540	.03663	.06475	-.02436	-.05630	.17540	.17540	-.28552
5.260	-0.8464	-.26890	.16110	.03301	.08414	-.03419	-.05738	.16110	.16110	-.26101
5.260	-0.95708	-.35636	.16820	.03270	.11583	-.04396	-.06010	.16920	.16920	-.24295
5.260	-1.0131	-.42626	.03736	-.00014	.00753	-.00270	-.00756	.00098	.00098	.00727

RUN NO. 0/ 0 RWAL = 5.24 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CT	CA	CLM	CYM	CSL	CL	CD	L/D
7.320	-0.374	-.05723	.26226	.16420	.02943	-.06904	.03317	-.05723	.16420	-.27843
7.320	-0.483	-.06213	.19088	.16036	.03216	-.04375	.02434	-.06213	.16036	-.33655
7.320	-0.5422	-.06244	.13650	.15494	.03230	-.04145	.01983	-.06244	.15494	-.37356
7.320	-0.5310	-.06375	.08026	.15093	.03296	-.02219	.00613	-.06375	.15093	-.40974
7.320	-0.632	-.06543	.03018	.14876	.03371	-.00943	.00163	-.06543	.14876	-.43748
7.320	-0.6332	-.01397	-.01397	.14749	.03254	.00737	-.00369	-.06332	.14749	-.43053
7.320	-0.6105	-.06195	-.06195	.14863	.03132	.01673	-.00709	-.06105	.14863	-.40426
7.320	-0.5948	-.10757	-.10757	.15456	.02682	-.01091	-.00948	-.05948	.15456	-.36703
7.320	-0.5912	-.16430	-.16430	.15981	.02974	.04897	-.00912	-.05912	.15981	-.33400
7.320	-0.471	-.21746	-.21746	.16312	.02920	.06382	-.00651	-.05312	.16312	-.30259
7.320	-0.477	-.27674	-.27674	.17127	.02628	.08426	-.03443	-.05127	.17127	-.26638
GRADIENT	.00057	-.02410	.00036	.00036	-.00036	.00620	-.00240	.00067	.00036	.00025

TABULATED SOURCE ARC 3.5-169, 1A1D  
 AVES 3.5-169 1A1D C9 T10 AT2 FLUME ON

DATE 05 DEC 75

PARAMETRIC DATA

ALPHA = .000 AIRLON = .000  
 ELEVON = .000 BOFLAP = .000  
 SPDRBK = .000 RUDDER = .000

REFERENCE DATA

SRFP = 2680.0000 98.FT. XRRP = 1076.4800 IN.  
 LREF = 1290.0000 IN. YRRP = .0000 IN.  
 BRFP = 936.8600 IN. ZRRP = 400.0000 IN.  
 SCALE = .0100

GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CY	CA	CLM	CYN	CBL	CL	CD	L/D
5.260	-9.593	-.06042	.26775	.16777	.03212	-.09160	.03417	-.06042	.16777	-.28764
5.260	-7.476	-.06036	.19641	.16341	.03605	-.06495	.02392	-.06036	.16341	-.35360
5.260	-5.407	-.06300	.13197	.15959	.03677	-.04373	.01538	-.06300	.15959	-.37944
5.260	-3.320	-.06323	.07171	.15682	.03675	-.02068	.00654	-.06323	.15682	-.39346
5.260	-1.675	-.06407	.01806	.15352	.03763	-.00185	-.00084	-.06407	.15352	-.41678
5.260	1.148	-.06675	-.02335	.15155	.03930	.00617	-.00268	-.06675	.15155	-.44529
5.260	2.268	-.06802	-.07430	.15090	.03952	.01730	-.00990	-.06802	.15090	-.47969
5.260	4.314	-.06862	-.13384	.14871	.03936	.03979	-.01466	-.06862	.14871	-.50708
5.260	6.440	-.05763	-.20008	.16646	.03263	.06387	-.02437	-.05763	.16646	-.50677
5.260	8.325	-.05828	-.26727	.17529	.03298	.08975	-.03409	-.05828	.17529	-.50515
5.260	10.489	-.05571	-.32708	.17574	.03045	.11424	-.04953	-.05571	.17574	-.50976
5.260	GRADIENT	.07722	-.02567	.00058	-.02702	.00715	-.02242	.00058	.00058	.00434

GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CY	CA	CLM	CYN	CBL	CL	CD	L/D
7.320	-9.580	-.05597	.25031	.16338	.02894	-.07670	.03115	-.05597	.16338	-.27575
7.320	-7.456	-.05820	.18795	.15826	.03058	-.05711	.02278	-.05820	.15826	-.32509
7.320	-5.399	-.05922	.12786	.15407	.03104	-.03662	.01421	-.05922	.15407	-.35873
7.320	-3.476	-.06042	.07585	.14999	.03178	-.01984	.00731	-.06042	.14999	-.39153
7.320	-1.567	-.06055	.02616	.14729	.03183	-.00481	.00162	-.06055	.14729	-.40928
7.320	1.159	-.06108	-.01339	.14607	.03163	.00488	-.00237	-.06108	.14607	-.41801
7.320	2.270	-.06034	-.05772	.14846	.03138	.01195	-.00491	-.06034	.14846	-.40257
7.320	4.307	-.05669	-.11107	.15285	.02943	.02870	-.01133	-.05669	.15285	-.35266
7.320	6.408	-.05662	-.16677	.15822	.02911	.04700	-.01913	-.05662	.15822	-.32020
7.320	8.489	-.05532	-.22103	.16443	.02741	.06512	-.02667	-.05532	.16443	-.28435
7.320	10.489	-.05330	-.27707	.17018	.02666	.08522	-.03472	-.05330	.17018	-.25375
7.320	GRADIENT	.00041	-.02359	.00038	-.00027	.00589	-.00041	.00038	.00038	.00465

MACH YACL  
 5.260 32.802892  
 7.320 14.902700



REFERENCE DATA

SREF = 2880.0000 86. FT. XGRP = 1076.4800 IN.  
 LREF = 1790.0000 IN. YGRP = .0000 IN.  
 BREF = 936.8800 IN. ZGRP = 400.0000 IN.  
 SCALE = .0100

RUN NO. 0/ 0 RVAL = 9.64 GRADIENT INTERVAL = -5.00/ 5.00

MAC	ALPHA	ON	CY	CA	CLM	CTN	CEL	CL	CD	L/D
5.300	-9.539	-28156	-.02866	.18946	.10685	.00736	-.01253	-.24827	.23350	-1.05469
5.300	-7.395	-23531	-.03017	.18267	.09608	.01054	-.00337	-.20985	.21144	-.98247
5.300	-5.365	-18837	-.03350	.17388	.08006	.01470	-.01826	-.16929	.19064	-.86798
5.300	-3.402	-13802	-.02742	.16474	.06229	.01014	-.00382	-.12800	.17264	-.74143
5.300	-1.510	-.09254	-.02394	.15962	.04566	.00743	-.00282	-.08441	.15800	-.59955
5.300	.175	-.04368	-.02341	.14848	.03178	.00760	-.00305	-.04831	.14831	-.36497
5.300	2.258	-.04039	-.02206	.14194	.01498	.00700	-.00301	-.00997	.14165	-.07738
5.300	4.354	.03999	-.02333	.13660	-.00002	.00849	-.00342	.02951	.13924	.21192
5.300	6.517	.08616	-.02126	.13323	-.01434	.00743	-.00317	.07048	.14215	.49580
5.300	8.547	.12475	-.01766	.12726	-.02596	.00541	-.00255	.10445	.14439	.72340
5.300	10.531	.16821	-.01365	.12136	-.04044	.00224	-.00125	.14900	.15003	.95319
GRADIENT		.02303	-.00031	-.00361	-.00805	-.00018	.00003	.02040	-.00427	.12460

RUN NO. 0/ 0 RVAL = 5.55 GRADIENT INTERVAL = -5.00/ 5.00

MAC	ALPHA	ON	CY	CA	CLM	CTN	CEL	CL	CD	L/D
7.320	-9.586	-26159	-.02893	.19065	.09310	.01137	-.00440	-.22369	.23451	-.96239
7.320	-7.396	-21882	-.02858	.18314	.08461	.00882	-.00346	-.19342	.20978	-.92201
7.320	-5.389	-17778	-.02774	.17453	.07221	.01053	-.00425	-.16060	.19045	-.84326
7.320	-3.418	-13841	-.02418	.16823	.05964	.00844	-.00364	-.12643	.17107	-.73976
7.320	-1.486	-.09476	-.02272	.15316	.04459	.00516	-.00447	-.09075	.15557	-.58338
7.320	.199	-.05938	-.02365	.14561	.03323	.00743	-.00334	-.05989	.14540	-.41187
7.320	2.285	-.02039	-.01549	.13776	.02332	.00181	-.00134	-.02607	.13683	-.19052
7.320	4.323	.01297	-.01614	.13309	.01473	.00385	-.00187	.00290	.13369	.02172
7.320	6.486	.04988	-.01649	.12885	.00648	.00281	-.00162	.03900	.13366	.26186
7.320	8.519	.08747	-.01587	.12231	-.00388	.00175	-.00113	.06838	.13391	.51066
7.320	10.541	.12861	-.01555	.11727	-.01540	.00385	-.00254	.10499	.13482	.75631
GRADIENT		.01934	.00124	-.00392	-.00575	-.00091	.00003	.01677	-.00482	.09932

MACN 5.300 7.320  
 CJAFTD .149138 .146459  
 CMAFTD -.057781 -.063408  
 CLMAFTD .033211 .034393  
 MAC/L -3.49371  
 YAC/L -297275

DATE 03 DEC 73

087011 ( 28 NOV 73 )

TABULATED SOURCE ARC 3.5-169, IA10  
AVES 3.5-169 IA10 C8 T10 AT2 FLUME OFF

REFERENCE DATA

SRFP = 2000.0000 SQ.FT. XMRP = 1076.4870 IN.  
LWRP = 1290.0000 IN. YMRP = .0000 IN.  
SRSP = 936.6820 IN. ZMRP = 400.0000 IN.  
SCALE = .0100

PARAMETRIC DATA

BETA = .000 AILRON = .000  
ELEVON = .000 BUFLAP = .000  
SPDRK = .000 RUDDER = .000

RNVL = 2.90 GRADIENT INTERVAL = -5.00/ 5.00

	ALPHA	ON	CY	CA	CLM	CVM	CEL	CL	CD	L/D
MACH = 5.260	-7.722	-23008	-01734	.21431	.10419	.00580	-.00214	-.20713	.24435	-.84767
	-5.757	-19111	-01280	.20318	.08821	.00439	-.00202	-.16976	.22133	-.76700
	-3.789	-14207	-01557	.19065	.07084	.00482	-.00197	-.12916	.19962	-.64702
	-2.465	-10784	-01574	.18241	.05688	.00460	-.00191	-.09990	.18688	-.53456
	-.066	-.04772	-01622	.16839	.03409	.00430	-.00173	-.04752	.16845	-.28213
	2.114	-.00098	-01091	.16100	.01817	.00217	-.00147	-.00692	.16285	-.04301
	4.171	.03956	-.00974	.15657	.00509	.00116	-.00120	.02807	.19903	.17680
	6.112	.06372	-.00693	.15098	-.00368	.00007	-.00112	.06717	.15904	.42277
	8.039	.12691	-01001	.14430	-.02448	.00098	-.00110	.10538	.16062	.65610
	9.774	.16953	-.00700	.13875	-.03965	-.00058	-.00075	.14351	.16552	.86754
	15.033	.29665	-.00307	.12865	-.06180	-.00078	-.00100	.23506	.20171	1.26448
	20.135	.43124	-.00052	.12032	-.12692	-.00036	-.00112	.36339	.26160	1.36914
	30.190	.76827	-.00344	.11322	-.26955	-.00254	-.00260	.60713	.48419	1.23390
GRADIENT	.02289	.00082	-.00434	-.00827	-.00827	-.00248	.00010	.01984	-.00317	.10458

MACH 5.260  
CAVFO .169085  
CMAFT -0.046180  
CLMFTI .033546  
XAC/L -0.361308

REFERENCE DATA

REF = 2690.0000 98.17. 30RP = 1076.4600 IN.  
 LREF = 1290.0000 IN. 14RP = .0000 IN.  
 SREF = 936.6000 IN. 24RP = 400.0000 IN.  
 SCALE = .0100

RVAL = 2.64 GRADIENT INTERVAL = -5.00/ 5.00

MAOH = 5.300

BETA	DOV/R	DCY/R	DCA/R	DCL/R	DCV/R	DOBL/R
-10.000	-.0011	.00093	-.00031	.00012	-.00065	.00021
-8.000	.0014	.00027	-.00090	-.00003	-.00045	.00012
-6.000	.0014	.00031	-.00044	-.00005	-.00043	.00011
-4.000	-.0002	-.00002	-.00046	.00000	-.00031	.00009
-2.000	.00021	.00063	-.00057	-.00007	-.00044	.00015
.000	.00003	.00043	-.00070	.00003	-.00036	.00013
2.000	.00013	.00072	-.00091	-.00014	-.00042	.00015
4.000	.00000	.00099	-.00093	-.00005	-.00036	.00018
6.000	-.00005	.00124	-.00124	-.00004	-.00061	.00025
8.000	.00010	.00203	-.00149	-.00011	-.00067	.00029
10.000	.00017	.00223	-.00160	-.00009	-.00067	.00033
GRADIENT	-.00000	.00010	-.00006	-.00001	-.00002	.00001

RVAL = 4.22 GRADIENT INTERVAL = -5.00/ 5.00

MAOH = 7.300

BETA	DOV/R	DCY/R	DCA/R	DCL/R	DCV/R	DOBL/R
-10.000	-.00006	.00033	-.00009	.00012	-.00032	.00017
-8.000	-.00033	.00025	-.00009	.00021	-.00026	.00013
-6.000	-.00027	.00064	-.00005	.00018	-.00042	.00017
-4.000	-.00039	.00012	-.00011	.00025	-.00016	.00009
-2.000	-.00033	.00026	-.00009	.00018	-.00009	.00007
.000	.00001	.00024	-.00015	.00008	-.00012	.00008
2.000	-.00016	.00000	-.00007	.00010	-.00003	.00003
4.000	-.00034	.00035	-.00019	.00019	-.00011	.00007
6.000	.00010	.00061	-.00023	.00002	-.00022	.00009
8.000	-.00018	.00030	-.00009	.00009	-.00016	.00007
10.000	-.00018	.00035	-.00009	.00014	-.00016	.00008
GRADIENT	.00001	.00002	-.00001	-.00001	.00001	-.00000

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR.

TABULATED SOURCE ARC 3.5-168, 1A10  
ANES 3.5-168 1A10 CB T10 AT2 PLUME ON

PARAMETRIC DATA

REFERENCE DATA

SWP = 2880.0000 SQ.FT. XGRP = 1078.4800 IN.  
LWP = 1290.0000 IN. YGRP = .0000 IN.  
SWP = 958.8800 IN. ZGRP = 400.0000 IN.  
SCALE = .0100

RM/L = 2.11 GRADIENT INTERVAL = -3.00/ 5.00

WACH = 5.300	DOV/DZ	DCY/DZ	DCA/DZ	DCL/DZ	DCV/DZ	DCBL/DZ
BETA						
-10.000	-.00228	.00137	.00082	.00018	-.00082	.00029
-8.000	-.00206	.00089	.00068	.00014	-.00057	.00025
-6.000	-.00019	.00095	.00072	.00014	-.00056	.00015
-4.000	-.00039	.00010	.00056	.00021	-.00018	.00008
-2.000	-.00030	-.00027	.00045	.00009	-.00000	.00002
.000	.00011	-.00102	.00053	-.00022	.00063	-.00044
2.000	.00003	-.00091	.00061	-.00011	.00077	-.00044
4.000	-.00015	-.00040	.00079	.00019	.00007	-.00008
6.000	.00007	-.00041	.00066	.00012	.00006	.00001
8.000	.00016	-.00033	.00102	-.00000	.00012	-.00003
10.000	-.00027	-.00067	.00125	.00015	.00012	-.00003
GRADIENT	.00004	-.00008	.00003	-.00001	.00006	-.00004

RM/L = 5.21 GRADIENT INTERVAL = -3.00/ 5.00

WACH = 7.300	DOV/DZ	DCY/DZ	DCA/DZ	DCL/DZ	DCV/DZ	DCBL/DZ
BETA						
-10.000	-.00009	.00140	.00005	.00002	-.00079	.00022
-8.000	-.00026	.00065	.00012	.00014	-.00047	.00015
-6.000	-.00033	.00082	.00010	.00014	-.00047	.00016
-4.000	-.00031	.00046	.00009	.00011	-.00028	.00010
-2.000	-.00048	.00025	.00013	.00018	-.00007	.00000
.000	-.00027	-.00005	.00015	.00010	.00023	-.00013
2.000	-.00006	-.00041	.00002	.00001	.00031	-.00023
4.000	-.00025	.00008	.00019	.00011	-.00008	-.00001
6.000	-.00025	.00011	.00011	.00007	-.00009	.00002
8.000	-.00033	.00028	.00007	.00015	-.00009	.00001
10.000	-.00033	.00003	.00012	.00018	-.00007	.00001
GRADIENT	.00003	-.00007	.00000	-.00001	.00005	-.00002