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NASA CR- 140362

FINAL PROGRAM REPORT

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DEVELOPMENT

OF A

BATTERY STATUS MONITOR

(NASA-CR-140362) DEVELOPMENT OF A BATTERY STATUS MONITOR Final Report (Chrysler Corp.) 46 p HC \$3.75 CSCL 14B $N75 - 12278$

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JULY 1974

FINAL PROGRAM REPORT FOR THE DEVELOPMENT OF A BATTERY STATUS MONITOR

JULY 1974

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APPROVED BY:

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FOREWORD

This report documents the results of the work performed by Chrysler Corporation Space Divisien (CCSD) on the development of a Battery Status Monitor under contract NAS 9-13654. This unit has the capability of providing the energy status of the battery, measure and transmit basic parameters, process those measurements **required** to determine abnormal functioning of the battery, and transmit warning signals of the abnormal condition along with a Go/No Go signal.

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SBCTION 1

INTRODUCTION

Chrysler Corporation Space Division has developed a prototype battery status monitor system for the Space Shuttle.

The Battery Status Monitor (BSM) functions to: 1) provide the energy status of the battery, 2) measure and transmit basic battery parameters, 3) process thjae measurements required to determine abnormal functioning by the battery, and 4) transmit warning signals of the abnormal condition along with a Go/No Go signal.

The objectives of the program were to design a system that would improve the design and operational characteristics of the Space Shuttle by:

- **o Improving turnaround capabilities**
- **o Lowering telemetry measurement requirements**
- **o Adding redundancy and autonomy**

To facilitate attainment of these objectives, the program was divided into the following phases:

- **o Phase I Technical Requirements Definition**
- **o Phase II Modifications for Alternate Battery**
- **o Phase III System Design**
- **o Phase IV System Performance Demonstration**

Section 2

EQUIPMENT REQUIREMENTS

2.1 General Requirements

Requirements for the BSM prototype unit were based on the following:

- o Must be capable of operating with a 10 amp-hour, 26 vented cell nickel cadmium (Ni-Cad) battery.
- o Must be capable of providing energy status (capacity) of the battery with proper compensation for charge efficiency, bat- , tery temperature and stand loss.
- o Must be capable of measuring and transmitting basic parameters on parallel continuous analog outputs.
- o Must be capable of processing various measured data to determine abnormal functioning of the battery.
- o Must be capable of transmitting warning signals of the above abnormal conditions along with a Go/No Go signal.
- o Must be capable of retaining memory of energy statue during extended stand periods.
- o Must be capable of accommodating silver cadmium or silver zinc batteries with minimum development cost and time.

`. 2.2 Battery Status Monitor

To meet the overall requirements the BSM has the following characteristics:

- o Power Input 28 plus or minus 4 volts DC. The energy status memory is supplied directly from the battery so as to be energized at all times.
- o Input Signals Capable of accepting the following input signals:

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• Energy status circuitry capable of compensating for the effect of the following parameters:

• Charge efficiency

• Charge efficiency with temperature compensation

- **Stand loss**
- **Capacity versus temperature**

• Data processed and the following signals transmitted:

o Data processed to determine abnormal functioning of battery and following warning signals transmitted:

Figure 2-1 Accuracy of Charge and Discharge Current

- GolNo Go signal transmitted based on state of combined warning signals.
- All signals transmitted by the BSM are 0 to S volts DC.

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Section 3

BOUI PMENT DESCRIPTION

3.1 Battery Status Monitor (BSM) System

The BSM system includes:

o battery Status Monitor Assembly (figure 3-1)

o Modified Battery Asses" (figure 3-2)

The interconnection of the above system is shown in figure 3-3. The connector pin functions are identified in table 3-1.

3.2 Battery Status Monitor Assembly

The BSM assembly consists of a power supply board and five wire wrap type electronic circuit boards (WT B). The schematic for the assembly is SKEE *Ii-b.* **The circuit boards are shown with the major components identified in figures 3-4 through 3-9. The board interconnection List is given by table 3-2. The adjustable functions are outlinad in table 3-3.**

The power board (PC1) converts the $28 + 4$ VDC input into the various voltages **required by the BSM. It consists of two commercially available DC-DG converters, which provide +15 VDC, -15 VDC, +12 VDC, -6 VDC and +5 VDC.**

The electronic circuit boards (PC2 through PC6) receive input power from PCI. Battery voltage, cell voltage, charge and discharge current, temperature and pressure signals are received from the modified battery assembly end converted to the following 0 to 5 volts output signals.

- **o Energy Status**
	- **Amp-Hour**
	- **Capacity**
- **o Current**
	- **Discharge Current**
	- **Charge Current**
- **o Median Temperature**
- **o Total Battery Voltage**

Figure 3-1. Battery Status Monitor Assembly

Figure 3-2. Modified Battery Assembly

Figure 3-3 System Interconnection Block Diagram

TABLE 3.1 CONNECTOR PIN CONNECTIONS AND FUNCTIONS

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TABLE 3.1 CONNECTOR FOR CONNECTIONS AND FUNCTIONS

Figure 3-4. PC-1 BSM Assembly Board

Figure 3-5. PC-2 BSM Assembly Board

Figure 3-6. PC-3 BSM Assembly Board

Figure 3-7. PC-4 BSM Assembly Board

Figure 3-8. PC-5 BSM Assembly Board

Figure 3-9. PC-6 BSM Assembly Board

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TABLE 3.2 PC BOARD INTERCONNECTION **LIST**

TABLE 3.2 PC BOARD INTERCONNECTION LIST

TABLK 3.2 PC BOARD **INTERCDNNECTION** LIST

* These pins are connected to the energy reset switch (ERB)

** These pins are connected to the relay used in stand loss circuit.

TABLE 3-3 PC CARD ADJUSTMENTS

TABLE 3-3 PC CARD ADJUSTMENTS

NOTE* All "A" type potentiometers are for sero adjustment of the amplifiers with the corresponding number of the potentiometer.

- Pressure
- Warning Signals
	- Overvoltage on charge
	- Undervoltage on discharge and on open circuit
	- Overtemperature
	- Overpressure
	- o Co /No Go

The charge and discharge current signals are converted to an amp-hour signal by applying the current input signals through a chopper stabilized amplifier to an integrator/pulse generator which emit pulses equivalent to a predetermined number of ampere-hours. The pulses are applied to an eight-bit up/down binary counter which also acts as a memory device during long stand periods to retain the output reading. The outputs of the counter are applied through inverters to a digital-to-analog (D/A) converter. The output of the D/A converter is signal conditioned and transmitted as the amp-hour output. Compensation for efficiency and stand loss is made at the integrator/pulse generator.

The amp-hour output signal is applied to amplifiers where it is temperature compensated to reflect battery capacity reduction due to temperature. The modified signal is transmitted as the capacity output.

Charge and discharge current signals from the battery are signal conditioned and transmitted as charge ind discharge current outputs.

Three temperature signals are received from the battery. The high and median temperatures are determined by logic circuitry. The median temperature signal is conditioned and transmitted as the median temperature output. The high temperature signal is compared to an overtemperature reference voltage and a five volt overtemperature warning signal is transmitted if the temperature signal exceeds the reference.

Battery voltage is measured, signal conditioned and transmitted as the total voltage output.

The cell voltages are received from the battery and applied through a resistance voltage divider network to the multiplexers. They scan the cell voltages and convert them to two single output cell voltage wave trains. The wave trains are applied through a differential amplifier to comparator circuits which monitor the cell voltages for under voltage during open circuit or discharge and overvoltage during charge. The output of the comparators are conditioned to provide undervoltage and overvoltage warning signals.

Battery case pressure is measured, signal conditioned and transmitted as the pressure output. The pressure output is also compared to a pressure reference voltage and a five volt overpressure warning signal transmitted when the signal exceeds the reference.

If any of the warning signals indicate an abnormal condition, a five volt No Go signal is transmitted. When all warning signals are in a zero state, a zero volt Go signal is transmitted.

3.3 Modified Battery Assembly

The test battery has been modified to provide the following information to the BSM assembly:

- **Charge/Discharge Current**
- **Battery Voltage**
- **Temperatures**
- **Pressure**

The modifications are as follows:

- **o Wiring to provide cell output voltages**
- **o Three linear thermistor networks mounted on three of the battery cell interconnection strips**
- **o A 100 amp shunt**
- **o A strain gage type pressure transducer mounted on the battery case.**
- **o Connectors to provide the necessary interconnection between the battery and BSM.**
- **3.4 Initial System Operation**

The BSM is initially mated with the modified battery as follows:

- **o Charge the battery to 100% and connect to BSM.**
- **o Apply power to the BSM.**
- **o Adjust the output of the BSM to 100% by depressing the energy reset switch (ERB) on the BSM assembly until the voltage across J2 pin A and J2 pin N reads S ± .04 VDC.**
- **o The BSM will now monitor battery status.**

Section 4

ACCEPTANCE TEST PERFORMANCE DATA

4.1 General

Results of the Acceptance Test are described in the following paragraphs. The test was run in accordance with the Acceptance Test Plan PL-EE-74-14.

4.2 Current Measurement

The charge current output signal was measured when the battery was being charged at 0.25 ampere, 0.50 ampere, 0.75 ampere and 1.0 amperes. A comparison of the test results with the expected results is given in table 4-1. All results were within the specified measurement test tolerance.

Table 4-1 Charge Current Test Results

The discharge current output signal was measured when the battery was discharging at 10 milliamps, 100 milliaaps, 1 ampere, 10 amperes and a simulated input signal of 100 amperes. A comparison of test results with the expected results is given in table 4-2.

Table 4-2 Discharge Current Test Results

Charge and discharge current percent errors are plotted for comparison with the test tolerances in figure 4-1.

The discharge current error is slightly greater than the specified tolerance because the tolerance represents the error at the output of the chopper stabi**lized** amplifier and does not take into account the output buffer amplifier cir**cuitry used to provide the 0 to 5 VDC telemetry output signal. The discharge current sig-ial errors** were adjusted to insure that the actual capacity of the **battery will be** always slightly greater than the readout value of capacity. The increase in the error at the 100 ampere test point is due to the use of simulation circuitry to provide **an** input signal rather than actually discharging the battery at 100 amperes.

4.3 Total Battery Voltage Measurement

The output was measured with battery input voltage at 0 volts, 19.97 volts, 30.05 volts and 43.73 volts. A comparison of results is given in table 4-3. All results were within the test **tolerance** of +1% between 20 and 45 VDC.

Voltage	Expected Value	Test Result	Error %
0.00 VDC	0,000 VDC	0,006 VDC	m
19,97 VDC	2.219 VDC	2,203 VDC	$-0.79%$
30.05 VDC	3,339 VDC	3.313 VDC	$-177%$
43.73 VDC	4.859 VDC	4.819 VDC	$-0.82%$

Table 4-3 Total Voltage Test Results

4.4 **Temperature Measurement**

the **three thermistor** networks were checked and found to work properly. The high and median temperature logic was checked for all combinations of high, median and low input temperature signals and worked properly. The median temperature owned was checked **at a** simulated 32°F, 80°F and 130**0F. Results are given in tabl-, --4.**

Table 4-4 Temperature Test Results

The error at 80 F is slightly above the test tolerance of $+1^{\circ}$ F. This is attributed to the imbalance between the circuitry used to simulate the three-thermistor input voltage network and slightly unmatched condition between the three temperature channel circuits used for median temperature selection. In actual operation.the therm istors are matched within +0.3**0F.**

Figure 4-1 Error Versus Current on BSM

4.5 Amp **Hour** Accounting

The asp-hour circuitry was tested without **current compensation for** discharges of **10 snperes** for 27 minutes and 20 **amperes** for 12 minutes. Comparison of test **re**sults with the expected values is given in table 4-5. Results were within the **test tolerance** of **+2%.**

Table 4-5 Amp-Hour Test Results

The amp-hour output was calibrated with temperature compensation adjustments made to the **charge accounting circuitry. The charge efficiency** was set for 98.8% at 12^{0} F, R9% at 78⁰F and 79% at 130⁰F.

The capacity output was tested over the temperature range of 32° F to 130° F. Results are given in table 4-6. All results were within the tolerance of $+5\%$.

Table 4-6 Capacity Test Results

The stand loss circuitry was **tested** and indicated a 12 +1.5% capacity correction over a ten day stand period.

4.6 Pressure Measurement

The **pressure** transducer was tested at **atmospheric pressure** and functioned properly. The pressure measurement circuitry was tested at atmospheric pressure and a pres- ' sure of 31 psia. Test results were within the tolerance of $+1$ psia and are given **in** table 4-7.

Table 4-7 Pressure Test Results

4.7 Action Signals (Warning Signals)

All warning signals were tested using simulated signals and functioned properly. **The Go/No Go signal was monitored during the test of the warning signals and it met the specified requirements.**

4.8 Impedance

Output impedance, insulation resistance and isolation resistance were measured and meet the tolerances given in the design/performance specification.

4.9 Supply Voltage

All outputs were measured with an input supply voltage of 24 volts and 32 volts with a noise level of 1.0 voltspeak to peak from zero to 20 KHz. All outputs **operated properly at the above supply voltages.**

4.10 Temperature

With the 8SM in a temperature chamber, operational checks were performed after a soak period of 30 minutes with the chamber set at 0° F, 130[°]F and 175[°]F. The opera**tional checks included the following:**

- **o Median temperature output at ambient.**
- **o Total battery voltage output.**
- **o Discharge current output with a discharge current of 10 amperes.**
- **o Charge current output with a charge current of 1 ampere.**
- **o Amp-hour and capacity circuitry.**
- **o Pressure output at atmospheric pressure.**
- **o Overvoltage, undervoltage and go/no go outputs with simulated abnormal conditions.**
- **o Overtemperature and overpressure outputs.**

All outputs operated properly at the three temperatures except the pressure out put. The pressure readout circuitry exhibited an output change with wide tempera-**; I:ure variations (approximately 9.7 mv/ F) which exceeds the specification requirement for thepressure circuit. Recommended improvement of this circuit is discussed in Section 5.**

4.11 Other Environmental Requirements Analysis

This analysis includes the operational capability and general modifications required on the BSM to meet the vibration, altitude and humidity environ and **re qui reagent • .**

4.11.1 Vibration

All the components used in the BSM have the capability to more than withstand the specified vibration requirement. However, the components would be repackaged **on P.C. cards using solder ratior than the wire wrap technique used in the prototype. In addition the integrated circuits would be soldered directly instead of** using plug-in sockets. The PC boards are mounted such that all four (4) sides of **the cards are supported resulting in prevention of high level mechanical resonances** that would result in overstressing.

4.11.2 Altitude

The unit should encounter no difficulties or present any operational hazard in operation under the required altitude environment. All voltages within the system are small (approximately 46 VDC max.).

4.11.3 Humidity

The SSM is a sealed unit. Therefore, no problem should be encountered during operation. ac the specified humidity environment. In addition the PC boards shall be coated in accordance with the material specification for the airborne equipsw,nt .

Section 5

RECOMMENDED IMPROVEMENTS

5.1 General

Although the system performs well, some desirable improvements were identified during the test program. Consideration should be given to incorporation of these improvements in equipment covered by follow-on programs. The following are recommended for consideration:

• Circuit Simplification and Component Reduction

• Amp-Hour Memory

• Discharge Current Range

• Pressure Circuit

• Total Power Reduction

5.2 Circuit Simplification and Component Reduction

Circuit simplification can be achieved both by review of present designed circuitry and use of IC components with improved circuit density (Quad Op Naps, Quad Comparators, etc.).

5.3 Amp Hour Circuitry Memory

The amp hour counter circuitry should be improved such that the counter will retain the battery status readout when input power is removed and reapplied. Reliable operation of the present system requires that 28 VDC input power be applied to the system cont4nuously. This improvement can be accomplished by either using a separate is*c*ilated power supply or by applying battery power to the amp-hour circuitry confinuously.

5.4 Discharge Current Ranges

Two five volt ranges of discharge current output can be used to improve the accuracy of this measurement. One range would correspond to discharges of 100 amperes to 1 ampere and the other range from 1 ampere to 10 milliamperes.

5.5 Pressure Circuit

The pressure measuring circuit should be redesigned to eliminate the variation of the pressure readout signal with large changes in temperature.

5.6 bower Dissipation

The amount of power being drawn from the battery should be reduced by increasing the impedance of the cell voltage divider circuitry, total battery voltage divi**der, and various other circuitry requiring battery power.**

Section 6

OTHER REPORTS

6.1 Reliability and Safety

The results of the reliability study are as follows:

The reliability on the originally defined mission (S components operating continuously for the, 7 day mission - 168 hrs., and the remaining components operate for 1 hour) yields a system reliability of .99966. This valve exceeds the goal of .999.

If the mission time was defined as all components having to operate for the entire 7 day mission (168 hrs) the total BSM system reliability would be only .98. In order to obtain the reliability goal of .999 for a mission of this length two entire BSM systems would have to be made redundant.

There are no single point failures in the BSM which could lead dir_ctly to damage of equipment or personnel.

6.2 Alcernate Battery Development Study

A study was made to determine the design changes and development cost in adopting the present BSM design to accommodate silver-cadmium or silver-zinc batteries. The characteristics considered in establishing the requirements included:

- **Cell and battery voltage**
- **Charge efficiency**
- **Temperature effect**
- **Charging methods**
- **Capacity lose characteristics**
- **1) Call and total battery voltage**

To accommodate the difference in number of cells and the per call voltage, the following circuits require revisions:

o The logic circuitry for the multiplexer to accommodate the increase or decrease in number of cells to be multiplexed.

- o The gain of the differential output amplifier for the cell voltage output circuitry.
- o Readjustment of the reference voltages for the "undervoltage" and "overvoltage" alarm circuitry.
- 2) Charge Efficiency

The charge efficiency circuitry would require readjustment.

3) Temperature Effects

The temperature compensation circuitry would require rescaling the various gains to accommodate silver-cadmium or silver-zinc batteries.

4) Charging

The charging method would require no change. However, the reference voltage levels for reducing the charger current levels would require rescaling to accommodate the voltage level characteristics of the silver-cadmium or silver-zinc battery.

5) Capacity loss characteristics

The capacity loss in temperature circuitry requires changes in gain and comparator reference voltages to provide for the output capacity profile for silver-cadmium or silver-zinc batteries.

The result of the study shows that the development of a universal type BSM that can accommodate nickel-cadmium and silver-cadmium or silver-zinc batteries will result in a per unit cost of approximately 12-15 per cent above the cost of the Ni-cad type BSM. The per unit cost of a silver-cadmium or silver-zinc unit would be the same as the Ni-cad units except for a small non-recurring cost for the design revision and documentation.

Appendix I

RELIABILITY/SAFETY STUDY ON BATTERY STATUS MONITOR (BSM)

FOR CONTRACT NAS9-13654

The results of a preliminary analysis of the Battery Status Monitor (BSM) are as follows:

Assessed board reliability on the originally defined mission **(5 components operating continuously for the** 7 day mission - 168 hrs., and :he remaining **components operate** for 1 hour) yields a system reliability of .99966. This **value exceeds the** goal of .999.

The reliability of each of the six boards is relatively close to one another. This is good in that there is no **one or** two "bad" boards i.e., boards with significantly higher failure rates which would result in repeated failures and high system down time.

The components with the highest failure rates are the D/A converter, and the three modular amplifiers Al, A9 and A10. The failure rates are not excessively high, however, and **in** fact are quite acceptable.

If the mission time was defined as all components having to operate for the entire 7 day mission (168 hrs) the total BSM system **reliability would be only** .98. In order to obtain the eliability goal of .999 for a mission of this length two entire BSM system, would have to be made redundant.

There are no single point failures in the BSM which could lead directly to damage to equipment or personnel.

Some background data are inc)-ded in the attached tables and diagram.

SYSTEM RELIABILITY BASED ON 2 DIFFERENT "ON" TIMES

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*1.32 is the failure rate of 5 components on board #2 which are supposed to be on for the entire 7 day mission $(t_2=168 \text{ hr})$. These components are BC-1C, BC-2C, R202, C₇, D₃.

Assuming all other components operate for t_1 =1 hr., then the system reliability 101

MEASUREMENT AND TOTAL SYSTEM RELIABILITY

BASED ON CONSTANT "ON" TIME

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MEASUREMENT FLOW DIAGRAM