

(NASA-CR-134908) HIGH-TEMPERATURE,
LOW-CYCLE FATIGUE OF COPPER-BASE ALLOYS FOR
ROCKET NOZZLES. PART 1: DATA SUMMARY FOR
MATERIALS TESTED IN PRIOR PROGRAMS
Technical Report, Jul. - (Mar-Test, Inc.,

N76-11273

Unclas
G3/26 01902

NASA CR-134908



HIGH-TEMPERATURE, LOW-CYCLE FATIGUE
OF COPPER-BASE ALLOYS FOR ROCKET
NOZZLES; PART I - DATA SUMMARY FOR
MATERIALS TESTED IN PRIOR PROGRAMS.

by: J.B. Conway, R.H. Stentz and J.T. Berling

MAR-TEST INC.
Cincinnati, Ohio
November, 1975
prepared for



NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

NASA Lewis Research Center
Contract NAS3-19720
G. R. Halford, Project Manager

1. Report No.		2. Government Accession No.		3. Recipient's Catalog No.	
4. Title and Subtitle High-Temperature, Low-Cycle Fatigue of Copper-Base Alloys for Rocket Nozzles; Part I-Data Summary for Materials Tested in Prior Programs				5. Report Date November, 1975	
				6. Performing Organization Code	
7. Author(s) J. B. Conway, R. H. Stentz and J. T. Berling				8. Performing Organization Report No. MTI-003-3-1	
				10. Work Unit No.	
9. Performing Organization Name and Address Mar-Test Inc. 45 Novner Drive Cincinnati, Ohio 45215				11. Contract or Grant No. NAS3-19720	
				13. Type of Report and Period Covered Contractor Report July through Oct., 1975	
12. Sponsoring Agency Name and Address National Aeronautics and Space Administration Lewis Research Center 21000 Brookpark Rd., Cleveland, Ohio 44135				14. Sponsoring Agency Code	
15. Supplementary Notes Project Manager, Dr. G. Halford, NASA-Lewis Research Center Cleveland, Ohio					
16. Abstract A more detailed analysis of the results obtained in 188 previously reported low-cycle fatigue tests of various candidate materials for regeneratively-cooled, reusable rocket nozzle liners is reported. Plots of load range versus cycles are reported for each test along with a stress-strain hysteresis loop near half-life. In addition, a summary table is provided to compare N_5 (cycles to a five percent load range drop) and N_f (cycles to complete specimen separation) values for each test.					
17. Key Words (Suggested by Author(s)) Low-Cycle Fatigue, Stress Range Data, Hysteresis Loops, High Temperature, Copper-base Alloys			18. Distribution Statement unclassified-unlimited		
19. Security Classif. (of this report) unclassified		20. Security Classif. (of this page) unclassified		21. No. of Pages	22. Price*

For sale by the National Technical Information Service, Springfield, Virginia 22151

TABLE OF CONTENTS

	<u>page</u>
I - SUMMARY.....	1
II - INTRODUCTION.....	2
III - MATERIAL AND SPECIMENS.....	4
IV - DISCUSSION OF DATA.....	6
a) Load Range Plots.....	9
b) Hysteresis Loops.....	198
c) Summary Table of N_5 and N_f	298
V - DISTRIBUTION LIST FOR THIS REPORT.....	304

I - SUMMARY

Five recent NASA Contractor Reports were written and published by Mar-Test Inc. to describe short-term tensile and low-cycle fatigue evaluations of eighteen (18) candidate materials (17 copper-base alloys along with pure silver) for regeneratively-cooled, re-usable rocket nozzle liners. Test temperatures ranged from room temperature to 593°C with all the elevated temperature tests being performed in high purity argon. All low-cycle fatigue tests were performed in axial strain control with strain ranges being selected to define the fatigue life to about 3000 cycles. A limited evaluation of strain-rate and hold-time effects was also included for selected alloys.

Following the formal publication of the results of these tests it was decided that a more detailed analysis should be made of each test to furnish a more comprehensive description of the low-cycle behavior of each material. This re-analysis focused on the 188 previously reported low-cycle fatigue tests and led to a plot of load range versus cycles for each test. It also yielded a typical hysteresis loop for each test chosen near the region of half-life. And, finally, this re-analysis led to a summary table to allow the values of N_5 (cycles to a five percent load range drop) and N_f (cycles to complete specimen separation) to be compared for each test.

In many of these tests a cyclic softening behavior was exhibited and it was not possible to identify a stabilized load range value for use in calculating N_5 . For these instances, the interpretation adopted was based on a five percent reduction from the load range value obtained at the point where a decided change in curvature was noted in the semi-logarithmic plot of load range versus cycles in the regime near fracture.

II - INTRODUCTION

In recent NASA programs, NAS3-16753 and NAS3-17777, a detailed evaluation was made of the short-term tensile and low-cycle fatigue behavior of numerous candidate materials for regeneratively-cooled, reusable rocket nozzle liners such as found in the engines of the Space Shuttle, Orbit-to-Orbit Shuttle, Space Tug, etc. The results obtained in these studies have been described in the following series of NASA Contractor Reports:

- NASA CR-121259 -- High Temperature, Low-cycle Fatigue of Copper-base Alloys in Argon; Part I - Preliminary Results for 12 Alloys at 1000⁰F (538⁰C).
- NASA CR-121260 -- High Temperature, Low-cycle Fatigue of Copper-base Alloys in Argon; Part II - Zirconium-copper at 482⁰C, 538⁰C and 593⁰C.
- NASA CR-121261 -- High Temperature, Low-cycle Fatigue of Copper-base Alloys in Argon; Part III - Zirconium-copper; Thermal-mechanical Strain Cycling, Hold-time and Notch Fatigue Results.
- NASA CR-134627 -- High Temperature, Low-cycle Fatigue of Advanced Copper-base Alloys for Rocket Nozzles; Part I - Narloy Z.
- NASA CR-134628 -- High Temperature, Low-cycle Fatigue of Advanced Copper-base Alloys for Rocket Nozzles; Part II - NASA 1.1, Glidcop, and Sputtered Copper Alloys.

Subsequent assessments by NASA personnel of the material property data presented in the above reports indicated that from a design point of view a more detailed evaluation of the stress-strain information recorded in the low-cycle fatigue tests was desirable. It was decided, therefore, that in Task I of this program a more comprehensive analysis would be made of each low-cycle fatigue test

performed during the last two NASA-sponsored programs performed by Mar-Test Inc. The strip chart records for each fatigue test were subjected to additional analysis so that a plot of load versus cycles could be prepared. In addition, it was decided that a typical hysteresis loop, selected near half-life, would be presented for each test. And finally, this Task I effort was to present a tabular summary for all the fatigue tests to compare the values of N_5 (cycles corresponding to a five percent load range drop) and N_f (cycles to complete specimen separation) for each test condition and material.

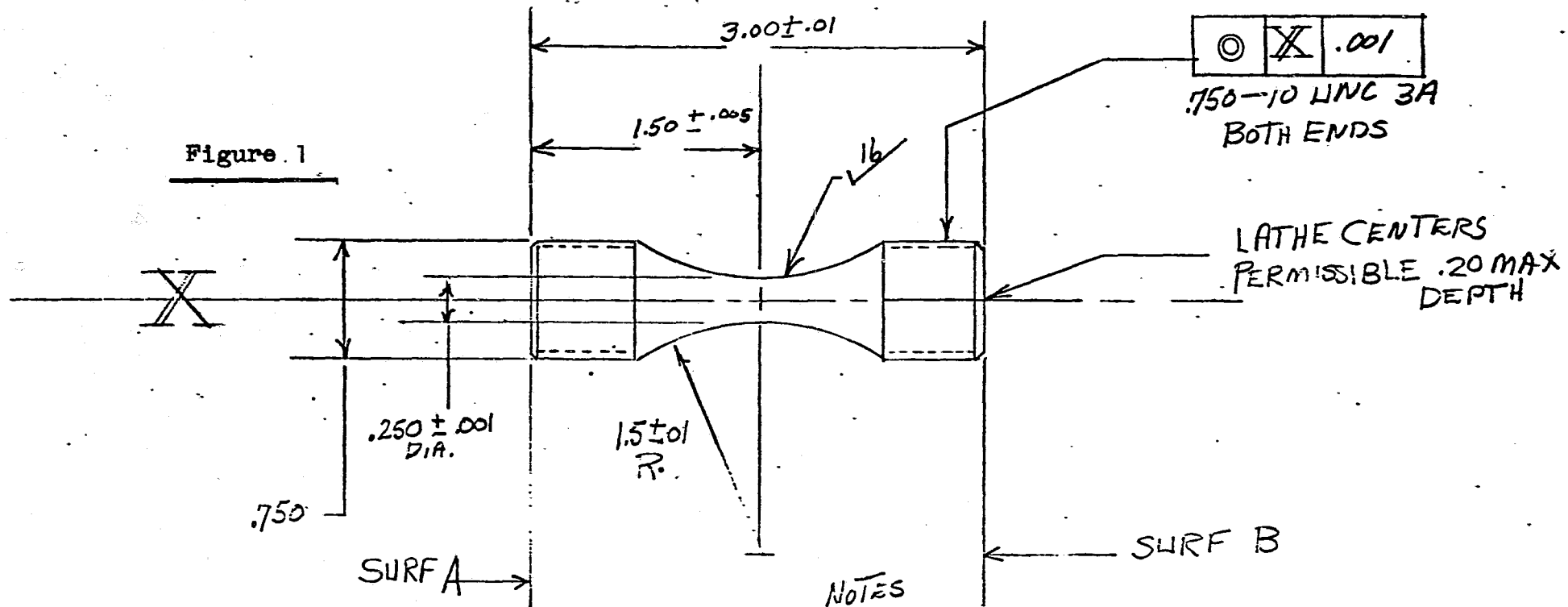
III - MATERIALS AND SPECIMENS

A total of nineteen (19) alloy designations was involved throughout the course of the material evaluations performed by Mar-Test Inc. on NASA Contracts NAS3-16753 and NAS3-17777. These designations together with a brief alloy definition are as follows:

R-0	Zirconium-copper, annealed
R-1	Zirconium-copper, $\frac{1}{4}$ hard
R-2	Zirconium-copper, $\frac{1}{2}$ hard
R-3	Tellurium-copper, $\frac{1}{2}$ hard
R-4	Chromium-copper, solution annealed and aged
R-5	OFHC copper, hard
R-6	OFHC copper, $\frac{1}{4}$ hard
R-7	OFHC copper, annealed
R-8	Silver, as-drawn
R-9	Zr-Cr-Mg copper, solution annealed, cold worked and aged
R-10	Electroformed copper, 30-35 ksi
R-13	Co-Be-Zr copper, solution annealed and aged
R-20	A second lot of zirconium-copper, $\frac{1}{2}$ hard
R-21	NASA 1-1A copper alloy, aged
R-22	NASA 1-1B copper alloy, as-received
R-23	Glidcop AL-10
R-24	Narloy Z, centrifugally cast, hot-rolled, solution annealed and aged
R-25	Sputtered zirconium-copper, annealed
R-26	Sputtered zirconium-copper, as-sputtered

All these materials were supplied by NASA-Lewis Research Center to Mar-Test Inc. and test specimens were then fabricated in accordance with the design shown in Figure 1. Special storage procedures were observed for the as-fabricated and post-test specimens and these are described in the contractor reports covering the test programs mentioned above.

Figure 1



5- SCREW THREADS TO BE AS LISTED IN NBS HANDBOOK H 28

NOTES

- 1- SURFACES A, & B TO BE PARALLEL WITHIN .001
- 2- SURFACES A, & B TO BE PERPENDICULAR TO CENTER LINE OF SPECIMEN WITHIN .0005 TIR
- 3- CONTOURED PORTION OF SPECIMEN TO HAVE A $\sqrt{16}$ FINISH OR BETTER. FINISHING SHOULD BE IN THE AXIAL DIRECTION USING LOW STRESS LAPPING OR POLISHING OPERATION
- 4- ALL DIA'S TO BE CONCENTRIC WITHIN .001 TIR

UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES TOLERANCES ON FRACTIONS DECIMALS ANGLES ± ± ±	DRAWN		SPECIMEN Low CYCLE FATIGUE	Mar-Test inc.	
	DATE			CINCINNATI, OHIO	
APPD		SIZE MTI-1002			
ISSUED					
MATERIAL	APPROVED	DATE	SCALE 1/1 WT CALC ACTUAL		
BOVT. OR COML. To BE SPECIFIED	ENGR <i>StC</i>	4-12-71			
	MFG				
	MATL				

IV - DISCUSSION OF DATA

A review of all the low-cycle fatigue tests performed on the two previous NASA-sponsored programs at Mar-Test Inc. led to the summary presented in Table 1. Each task of each program is treated individually to indicate the number of low-cycle fatigue tests performed for each material. This sequencing was also used in the presentation of the data plots and summary tables for this report. In other words, the data for the R-0 tests are presented first, followed by the data for the R-1 tests, etc. It will be noted that there are 188 tests involved in this data reporting activity.

For each test the load traces were reviewed first and a plot was made of the load range as a function of the number of cycles. In general, the first and second cycle information was read from the x-y traces (stress-strain hysteresis loops) and these points were plotted. Then this plotting was continued for subsequent cycles using load range information as read from the strip-chart recordings. A sufficient number of data points was selected throughout the fatigue life to define a fairly smooth pattern for the plot of load range versus cycles. Such plots for the 188 tests involved in this evaluation are presented in Figures 2 through 189. Each figure identifies alloy code number (see Table 1), specimen number, test temperature, strain range, strain rate and cycles to failure.

A typical hysteresis loop for each test is presented in Figures 190 through 288. It was the intent here to show the cyclic stress-strain behavior near half-life but in many cases no hysteresis loops were recorded in the immediate vicinity of $N_f/2$. In these instances the hysteresis loop selected for reporting was the one closest to half-life and usually in the region prior to $N_f/2$. The particular cycle used in such plots is recorded in the figure along with the temperature, strain range and cycles to failure.

A final data summary prepared for this report is presented in Table 2. For each test, values of N_5 , the cycles to a five percent load range drop, and N_f , cycles to complete specimen separation, are compared. Some interpretation of N_5 had to be made in those instances where cyclic softening was exhibited. In these situations the plot of load range versus cycles on semi-logarithmic coordinates indicated a gradual decrease in the load range as cycling progressed and it was not possible to identify a stabilized load range for calculating a five percent reduction. The special interpretation adopted for these cases involved the selection of that point on the load range plot where a change in curvature first began to appear as the load range began to decrease rapidly as failure approached. A five percent drop from this load range value was then used to establish the value for N_5 . In a few cases no value of N_5 is reported (in Table 2) due either to the fact that the specimen failed before a five percent load range reduction was reached or, because of the orientation of the crack with respect to the extensometer tips, the load range actually increased slightly near the failure point.

Table 1 - Summary of Low-Cycle Fatigue Tests Performed by Mar-Test Inc.
in Two Previous Programs

CONTRACT NAS3-16753; initiation date: June 29, 1972

<u>Alloy Designation</u>	<u>Material</u>	<u>Number of Tests</u>
Task I *****		
R-0	Zirconium-copper, annealed	6
R-1	Zirconium-copper, $\frac{3}{4}$ hard	6
R-2	Zirconium-copper, $\frac{1}{2}$ hard	8
R-3	Tellurium-copper, $\frac{1}{2}$ hard	6
R-4	Chromium-copper, SA and aged	6
R-5	OFHC copper, hard	6
R-6	OFHC copper, $\frac{1}{4}$ hard	7
R-7	OFHC copper, annealed	6
R-8	Silver, as-drawn	6
R-9	Zr-Cr-Mg copper, SA, CW and aged	7
R-10	Electroformed copper, 30-35 ksi	6
R-13	Co-Be-Zr copper, SA and aged	6
Task II *****		
R-2	Zirconium-copper, $\frac{1}{2}$ hard	25
Task III *****		
R-2	Zirconium-copper, $\frac{1}{2}$ hard	17
R-20	Zirconium-copper, $\frac{1}{2}$ hard	6

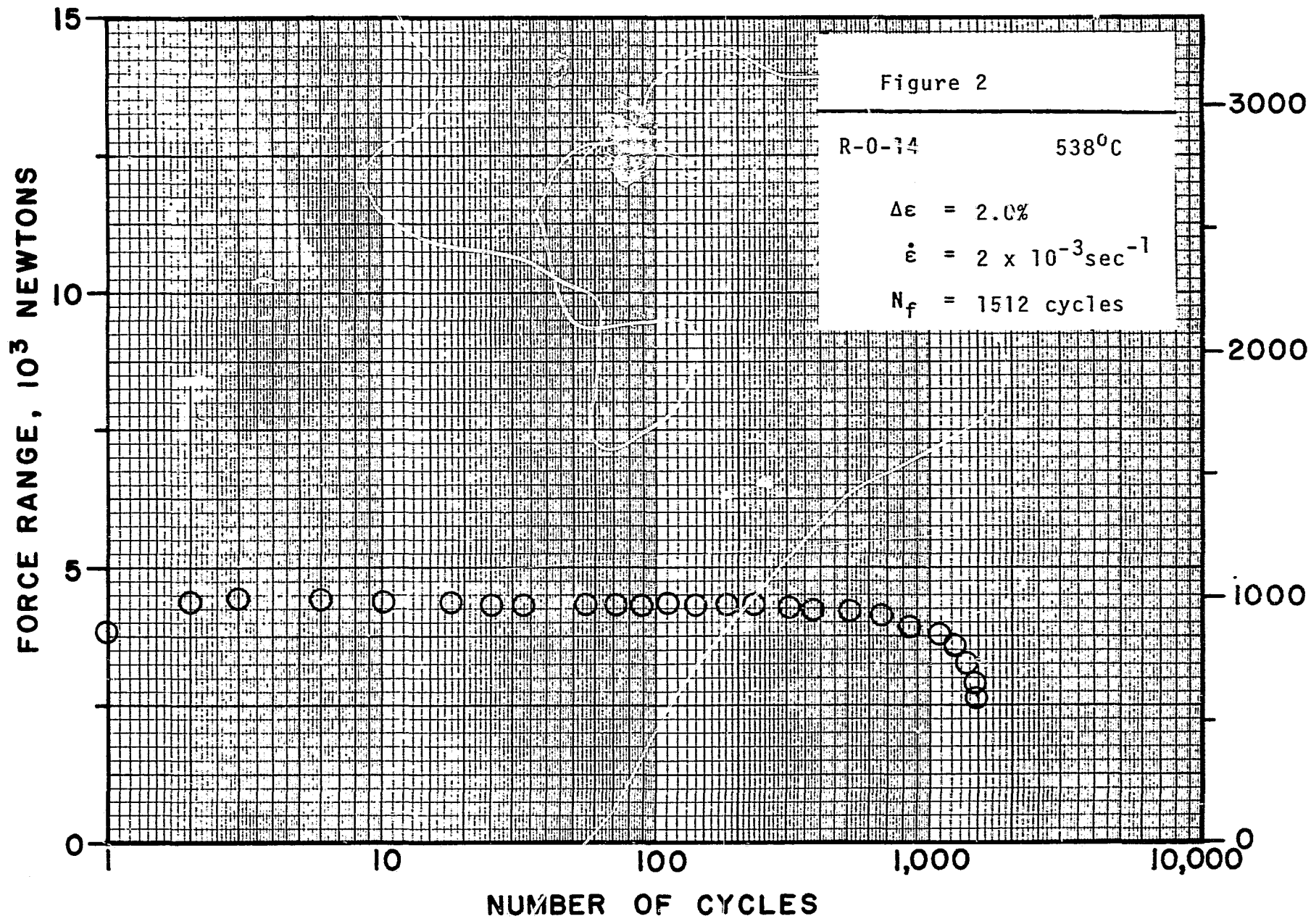
Table 1 continued - Summary of Low-Cycle Fatigue Tests Performed by Mar-Test Inc. in Two Previous Programs

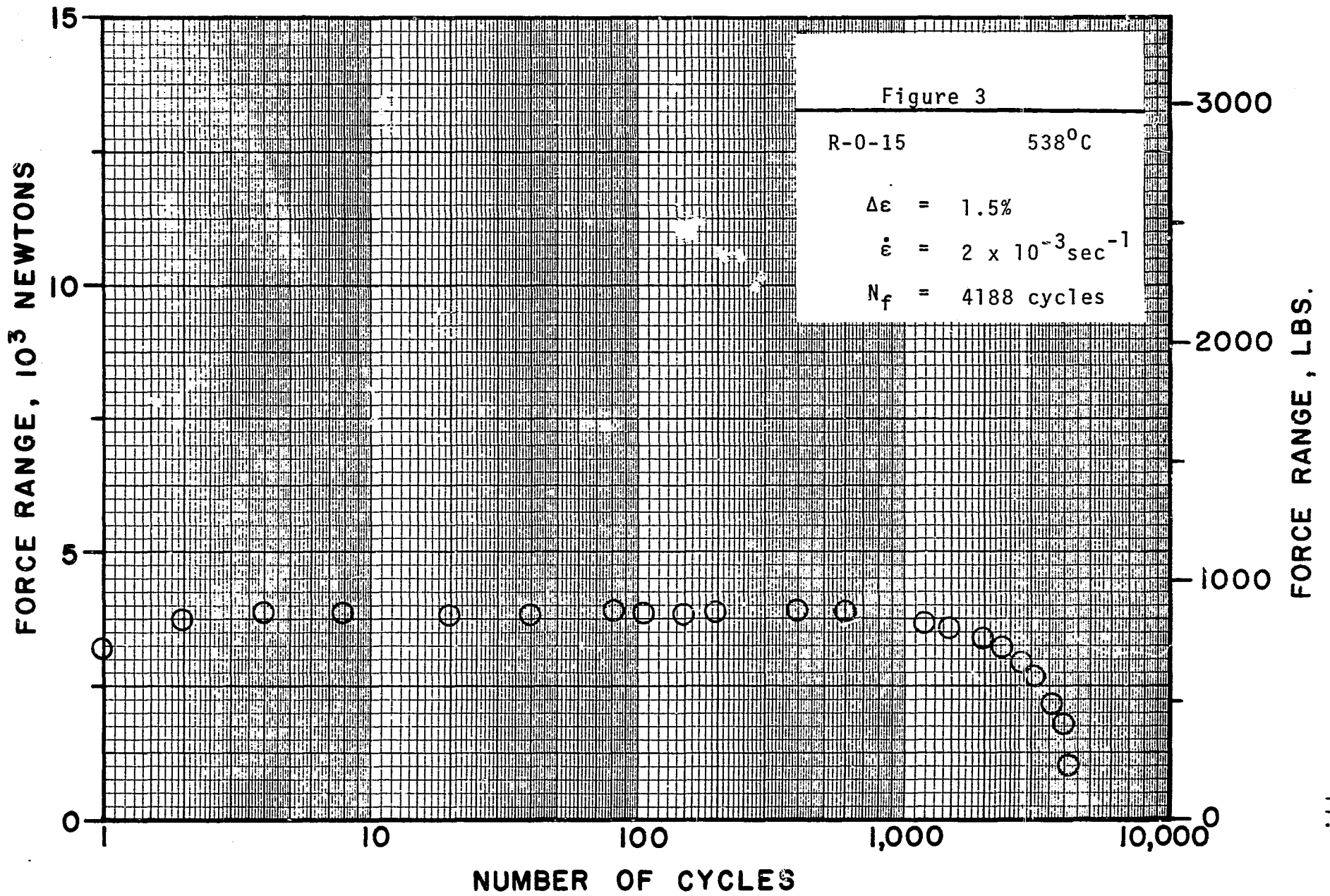
CONTRACT NAS3-17777; initiation date: June 28, 1973

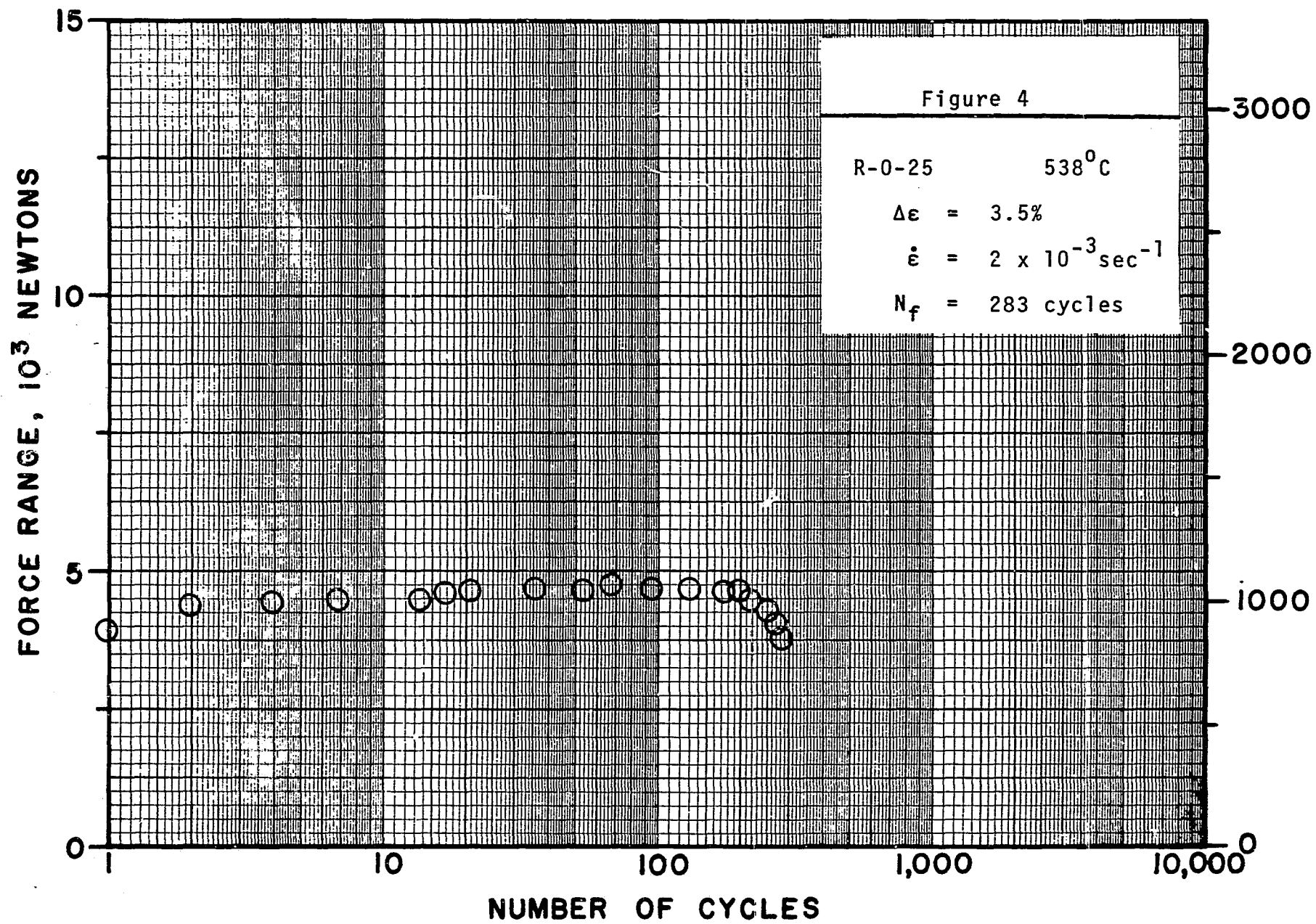
<u>Alloy Designation</u>	<u>Material</u>	<u>Number of Tests</u>
Task I *****		
R-24	Narloy Z, HR, SA and aged	29
R-2	Zirconium-copper, 1/2 hard	2
Task II *****		
R-21	NASA 1-1A copper alloy, aged	4
R-22	NASA 1-1B copper alloy, as-received	20
R-23	Glidcop AL-10	4
R-25	Sputtered zirconium-copper, annealed	4
R-26	Sputtered zirconium-copper, as-sputtered	1
Overall Total:		188 tests

a) LOAD RANGE PLOTS

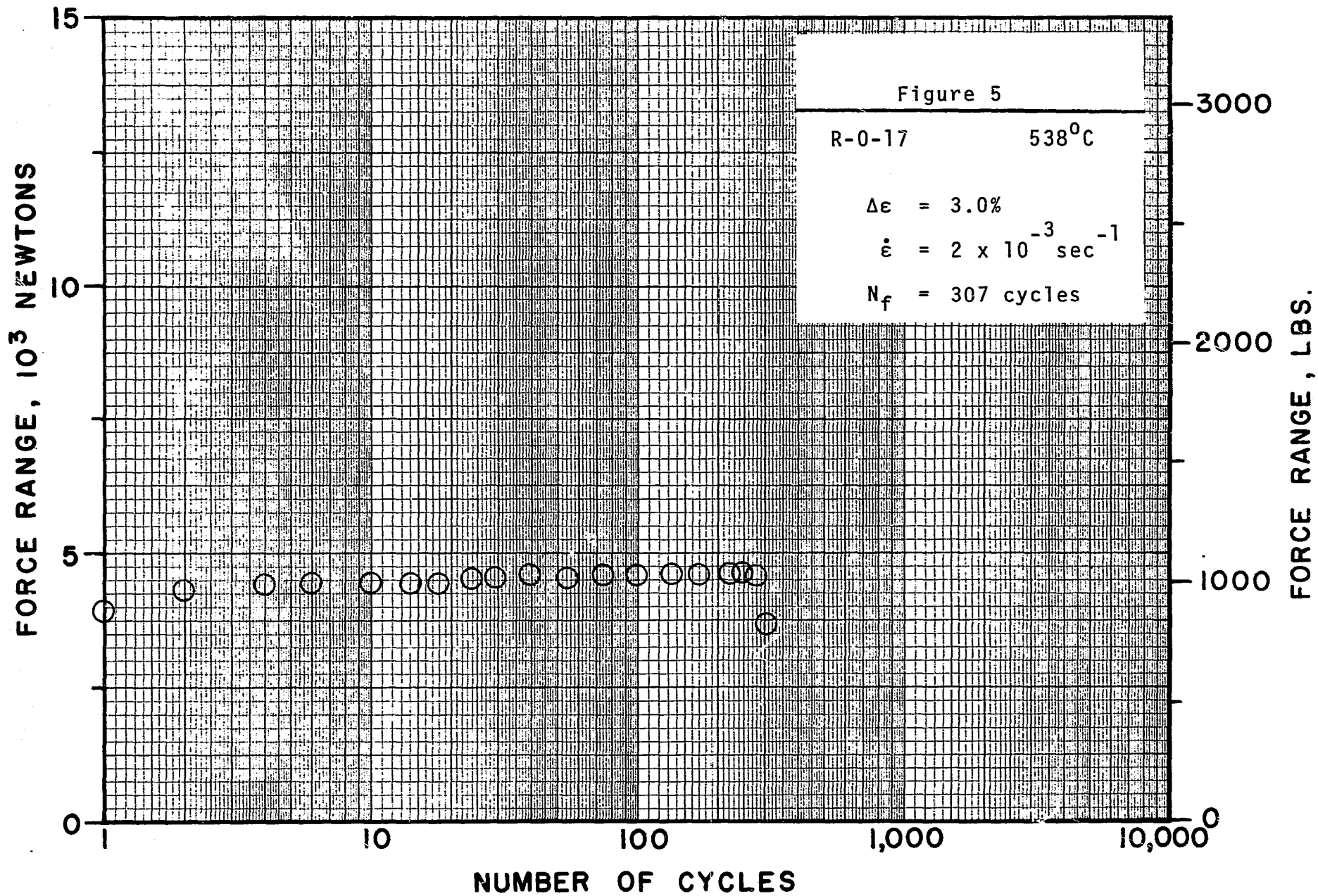
Figures 2 through 189

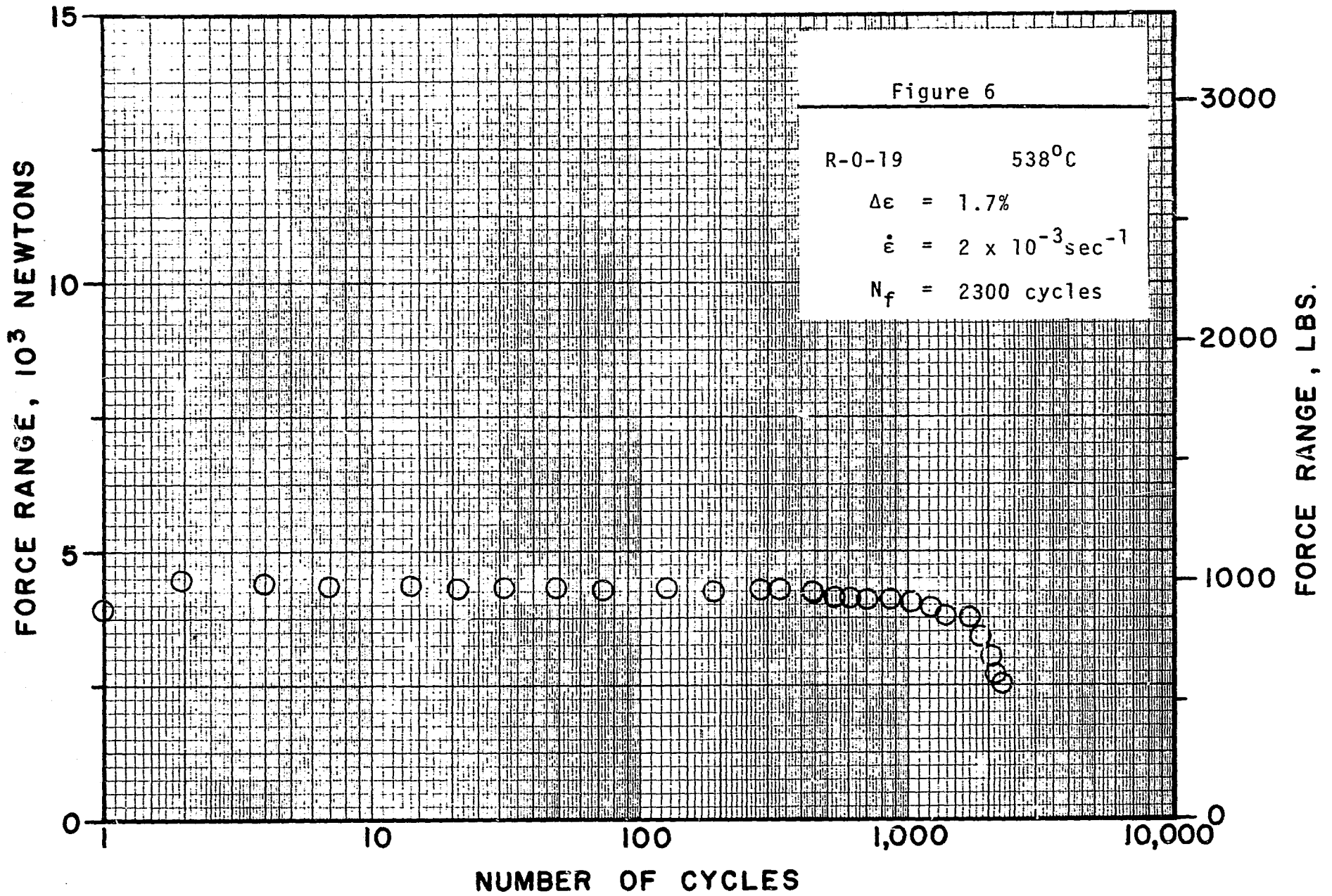


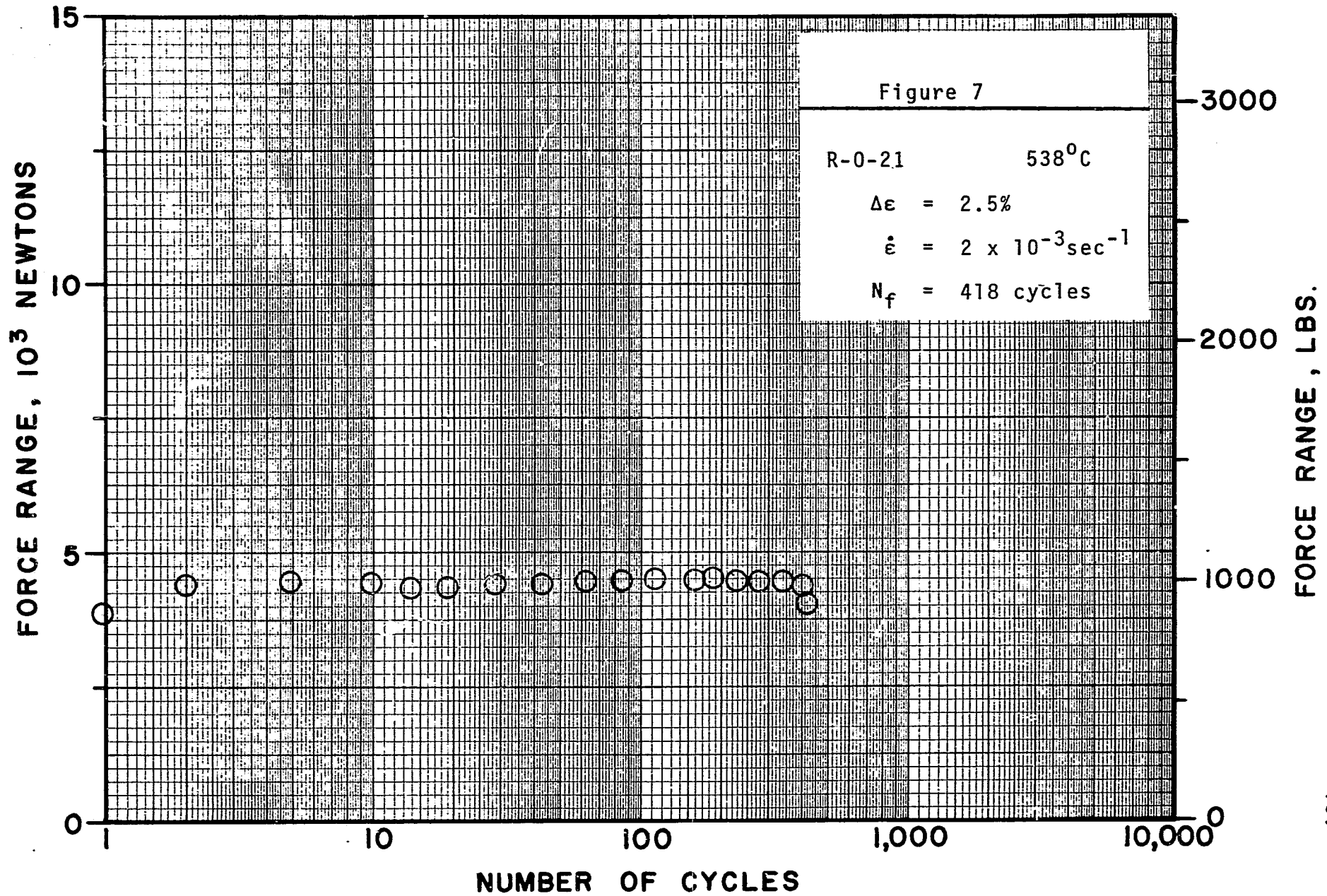


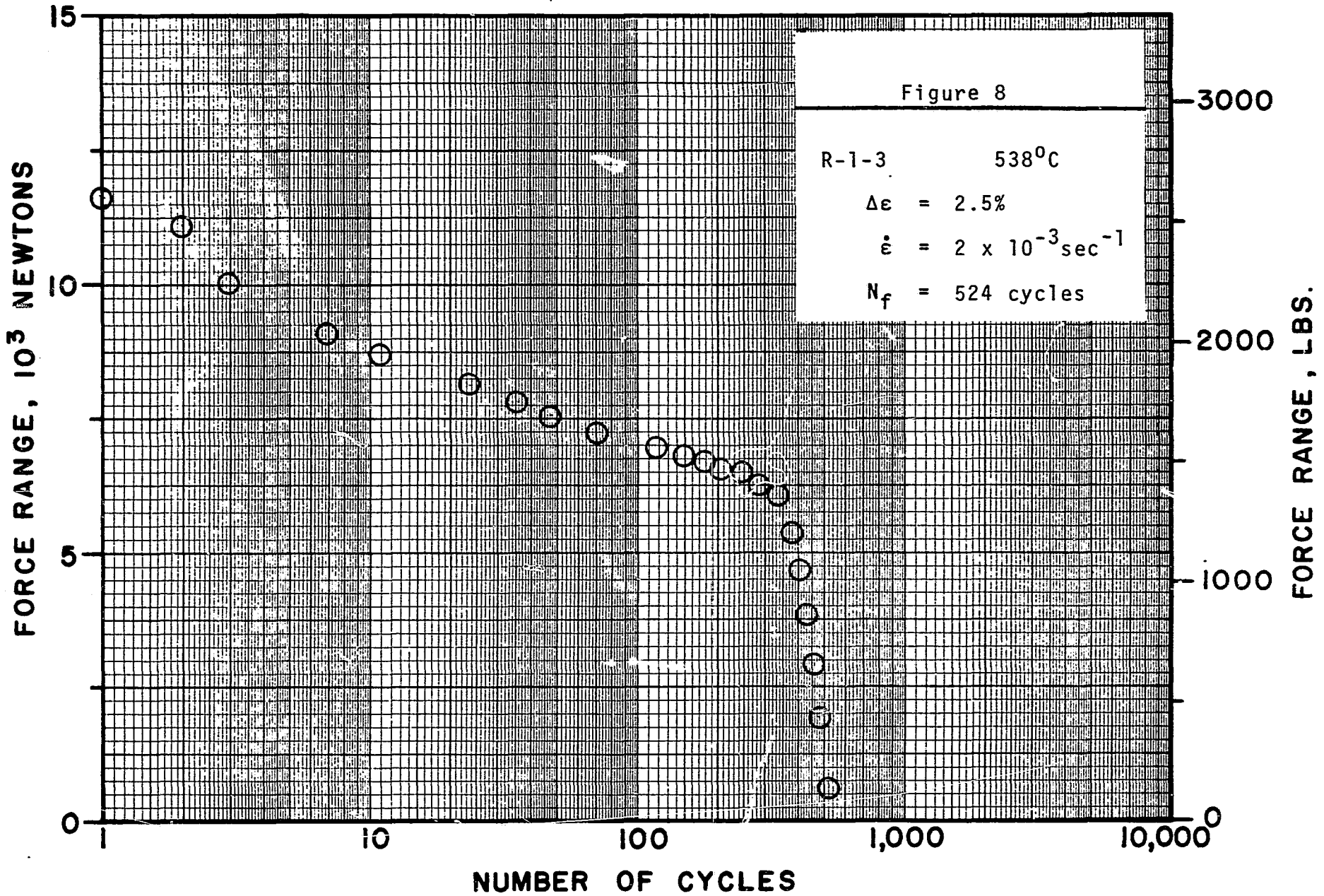


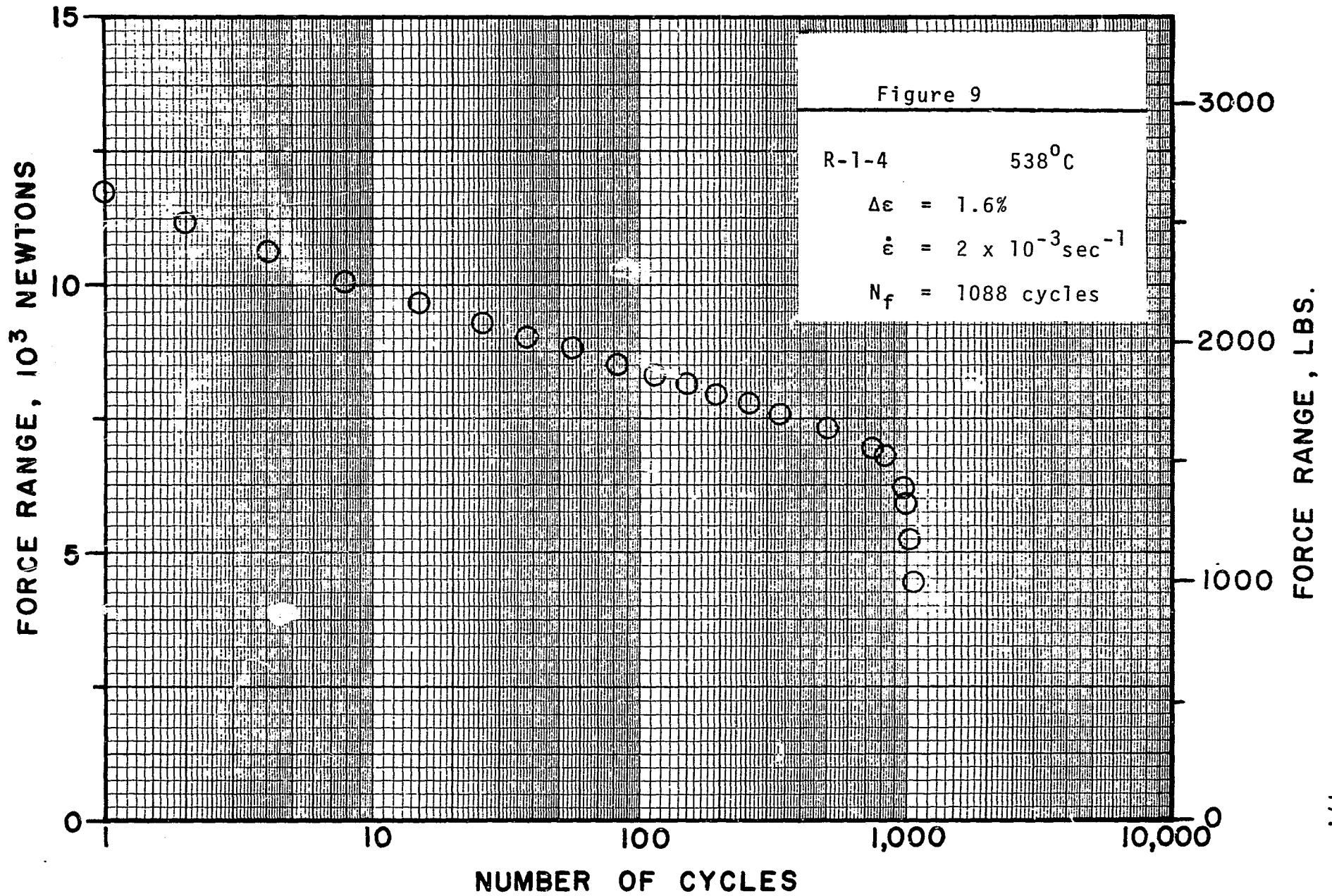
FORCE RANGE, LBS.

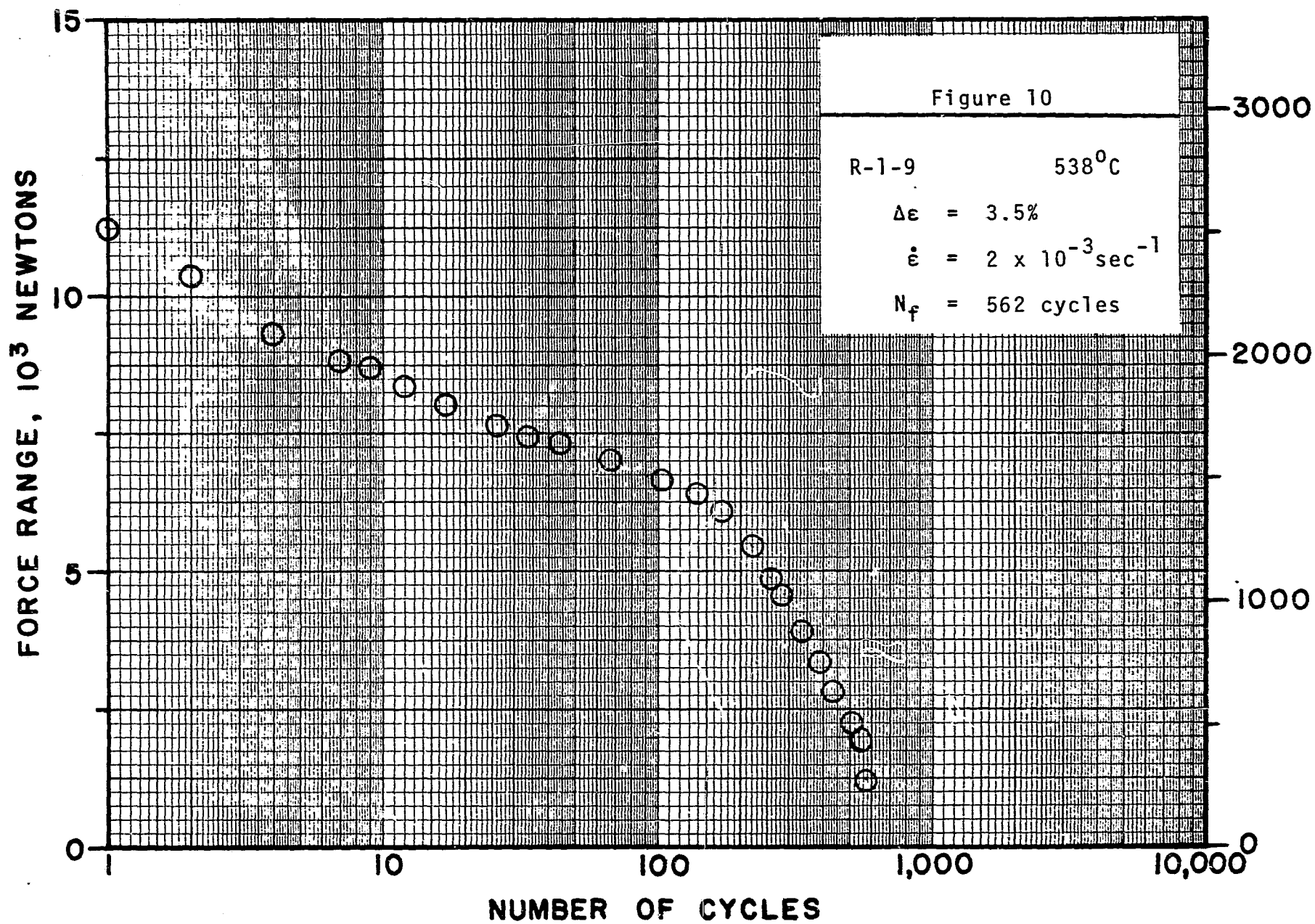


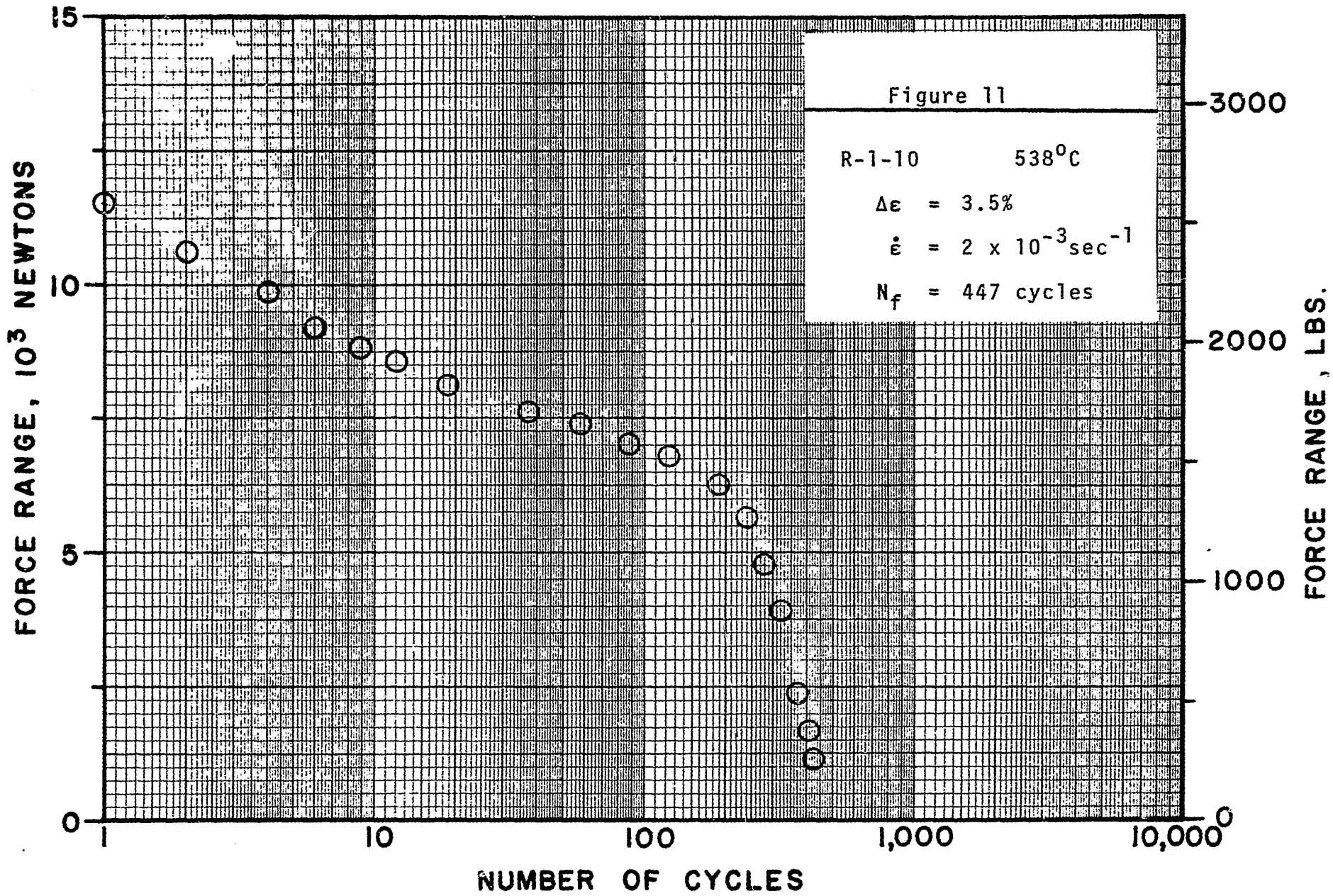


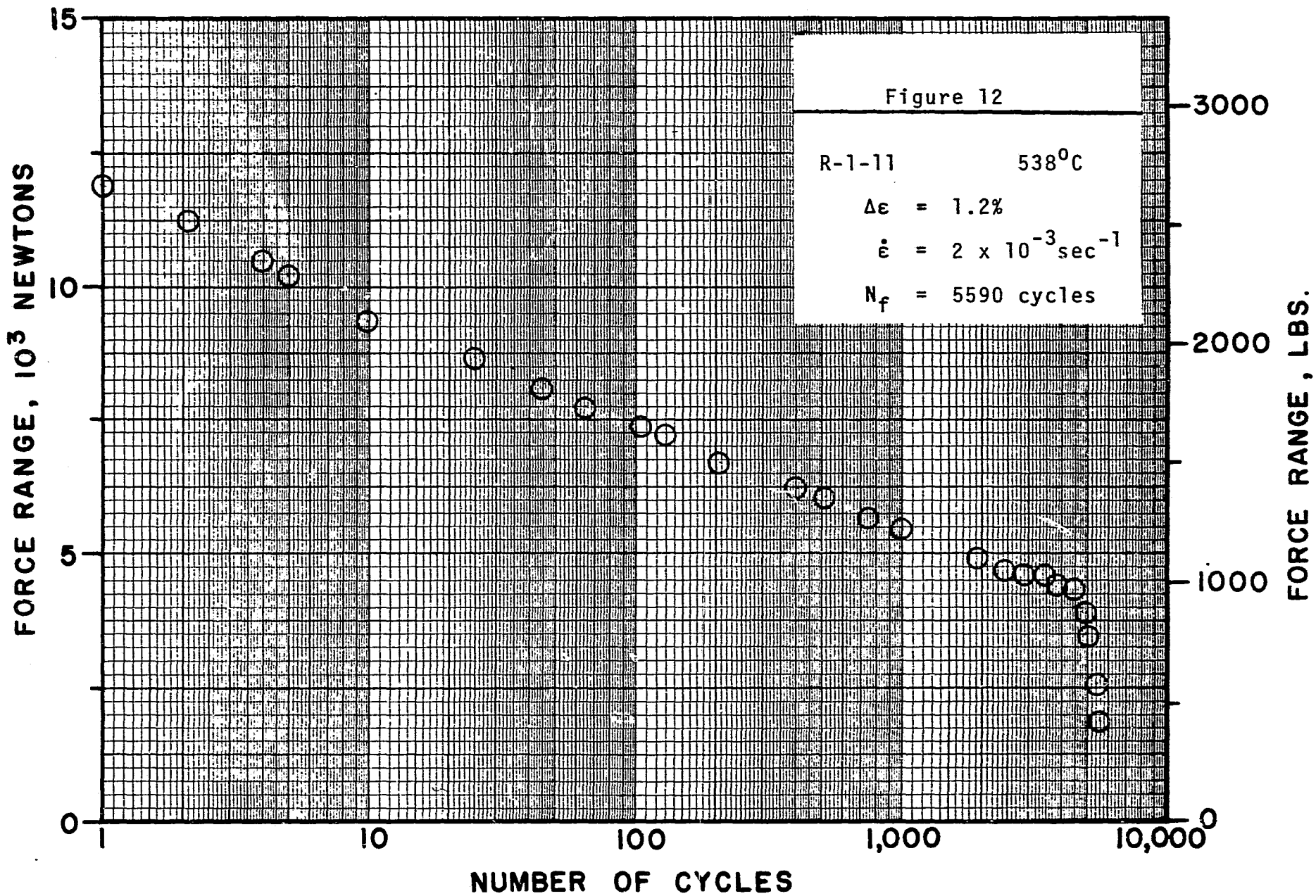


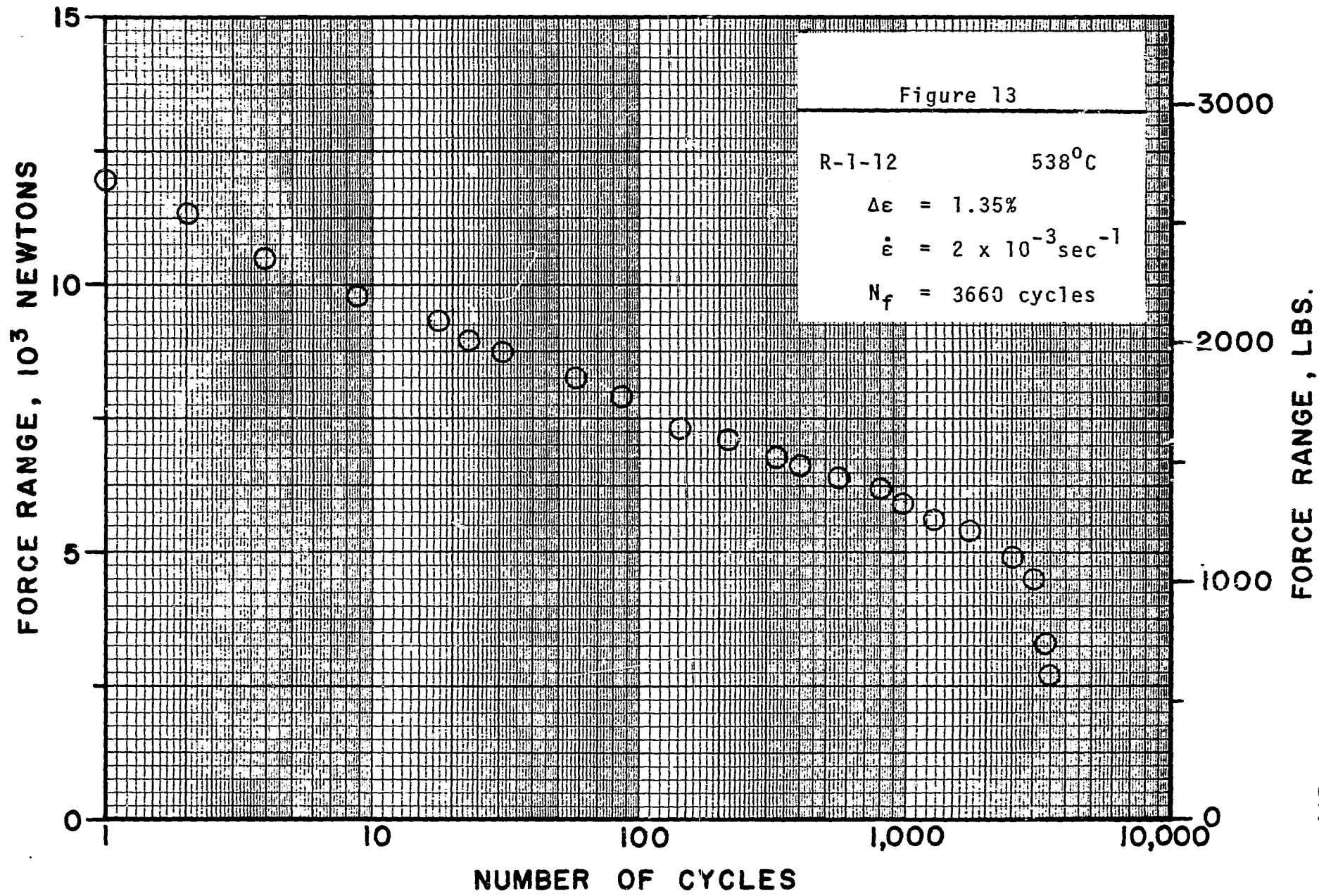


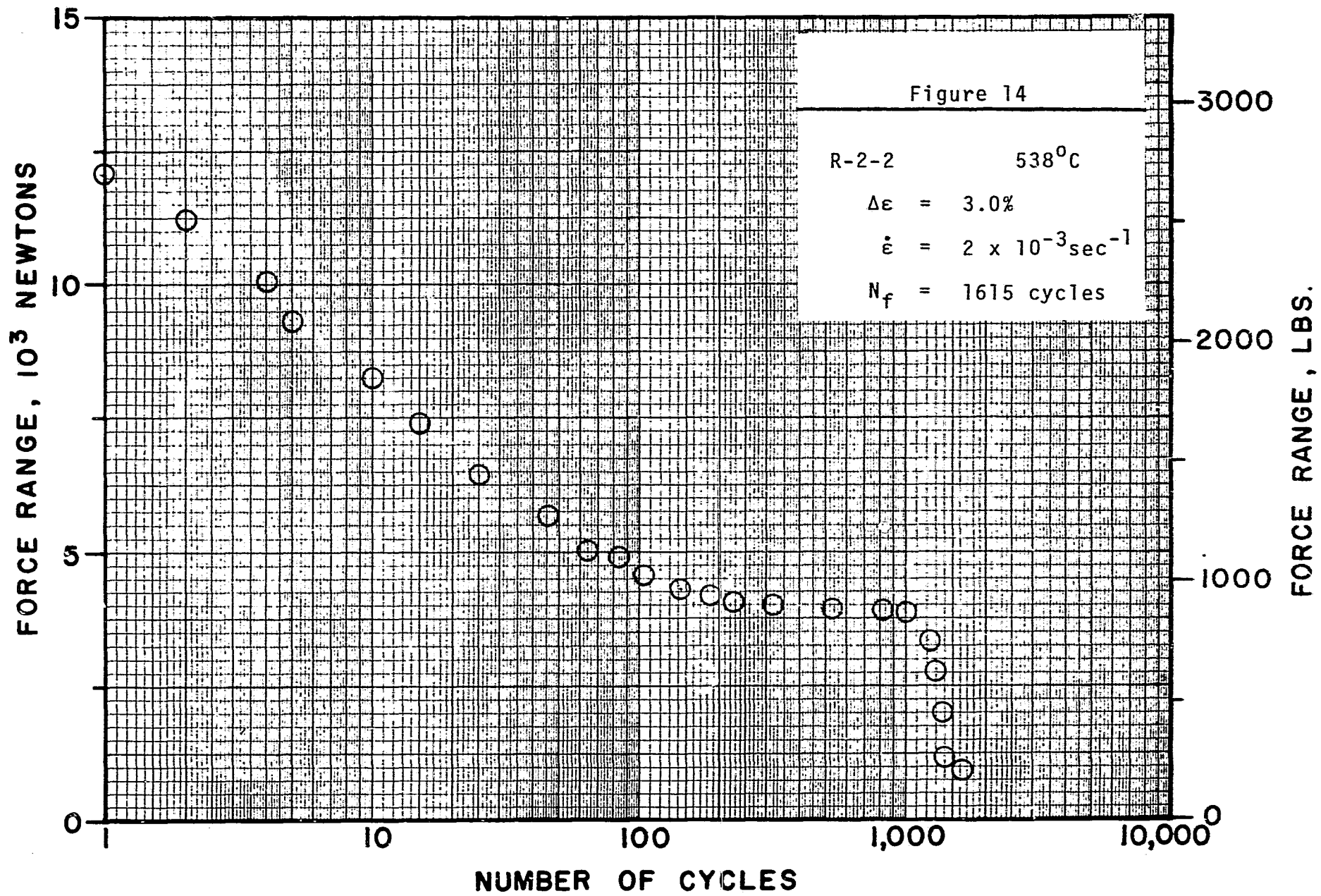


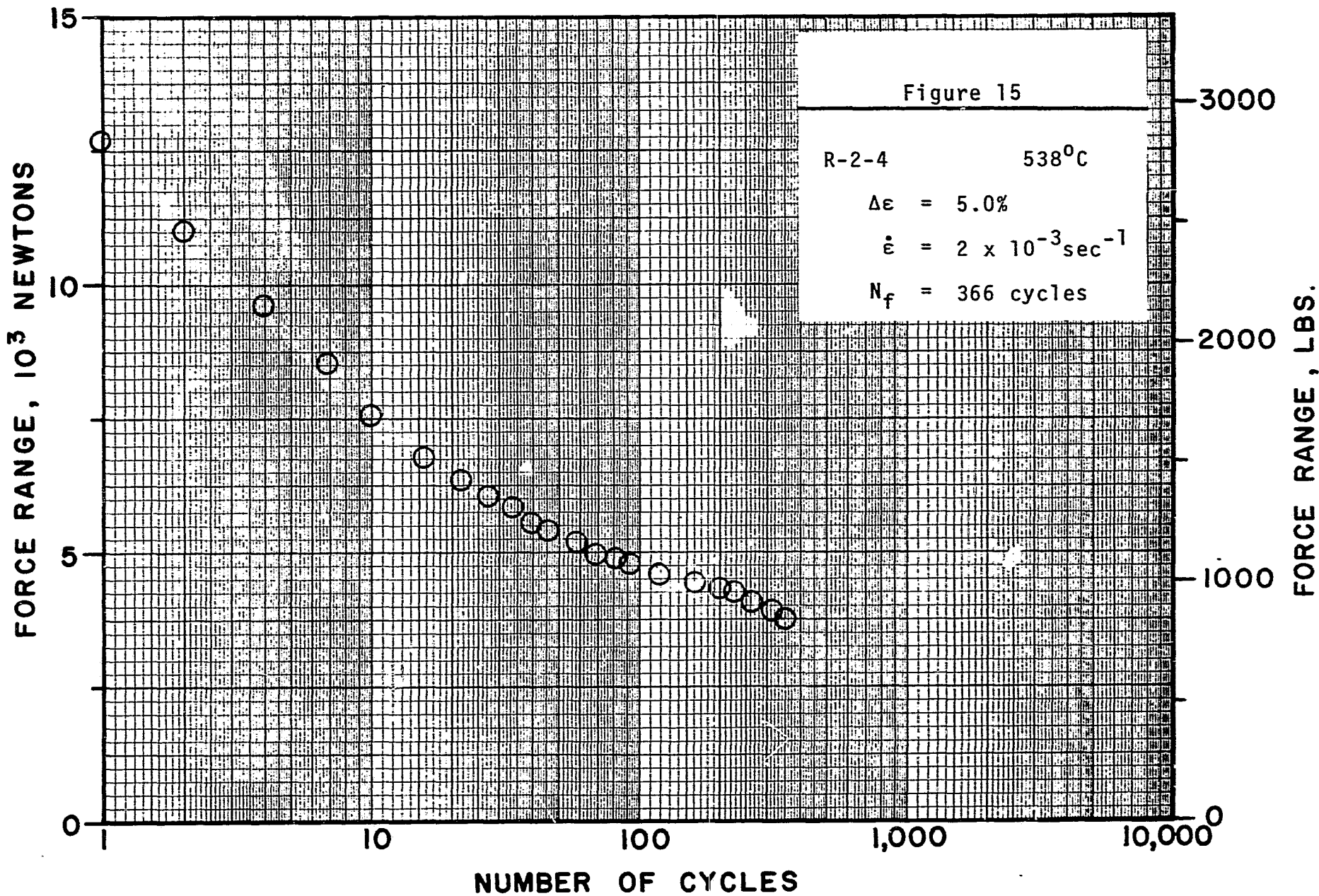


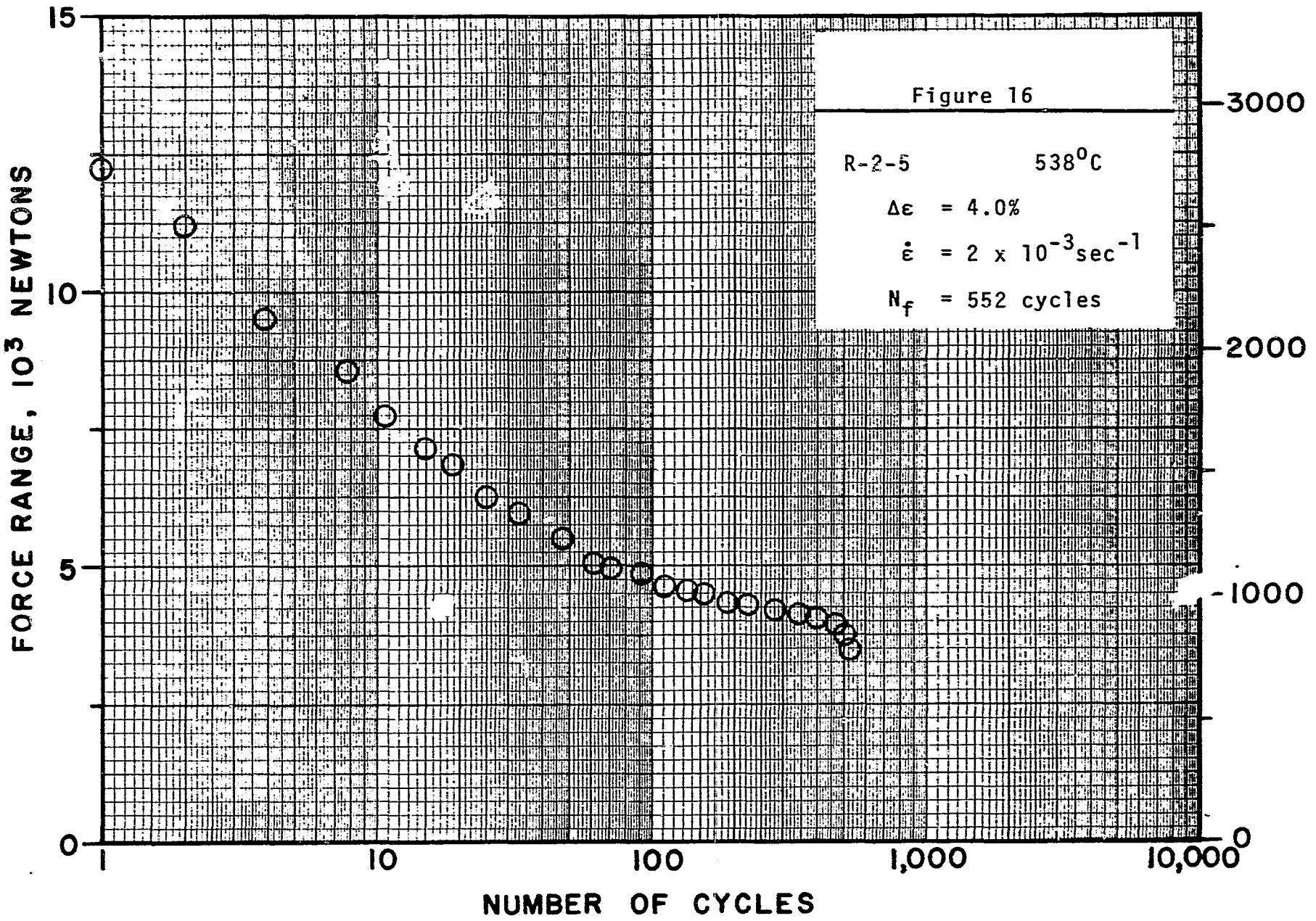


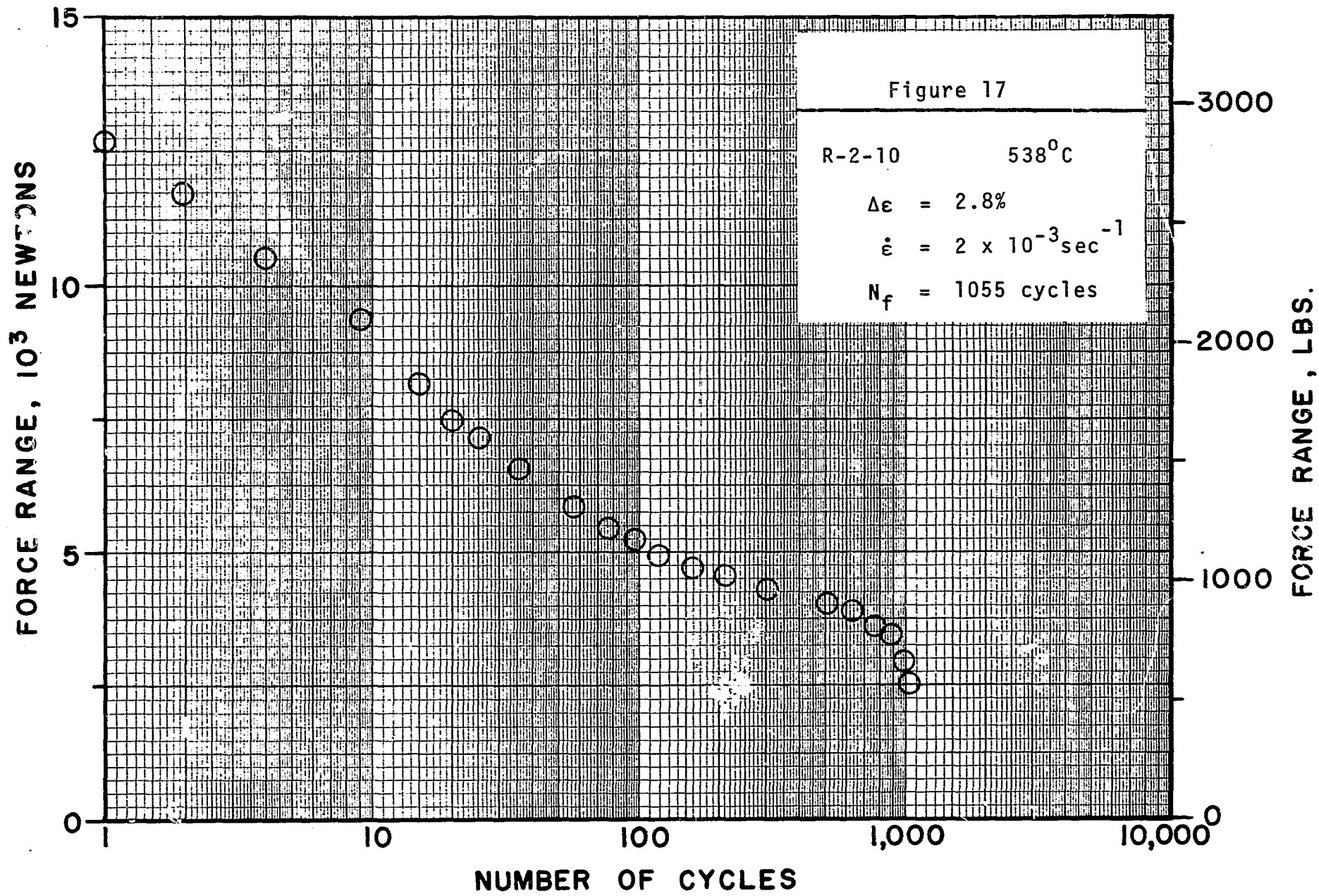


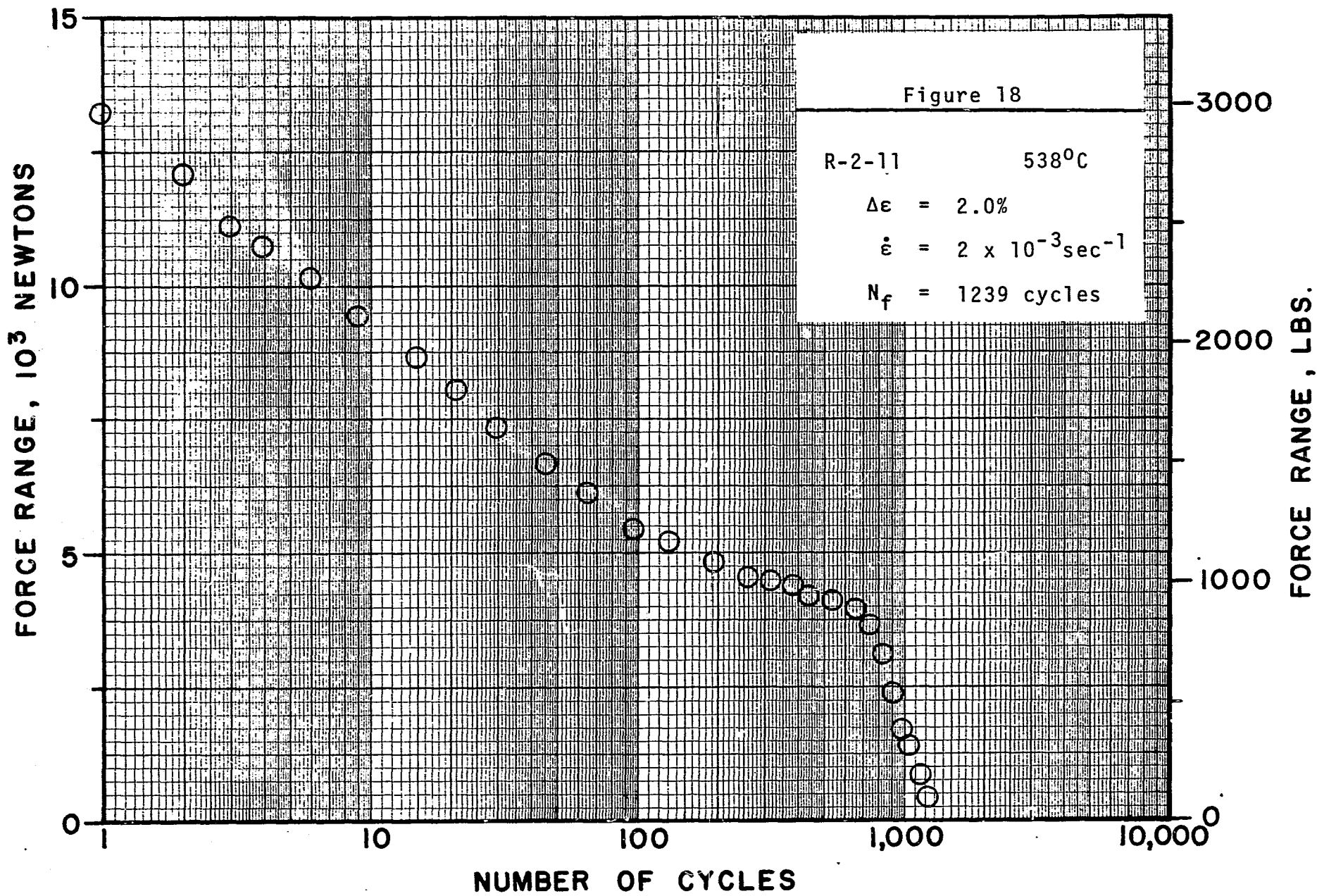


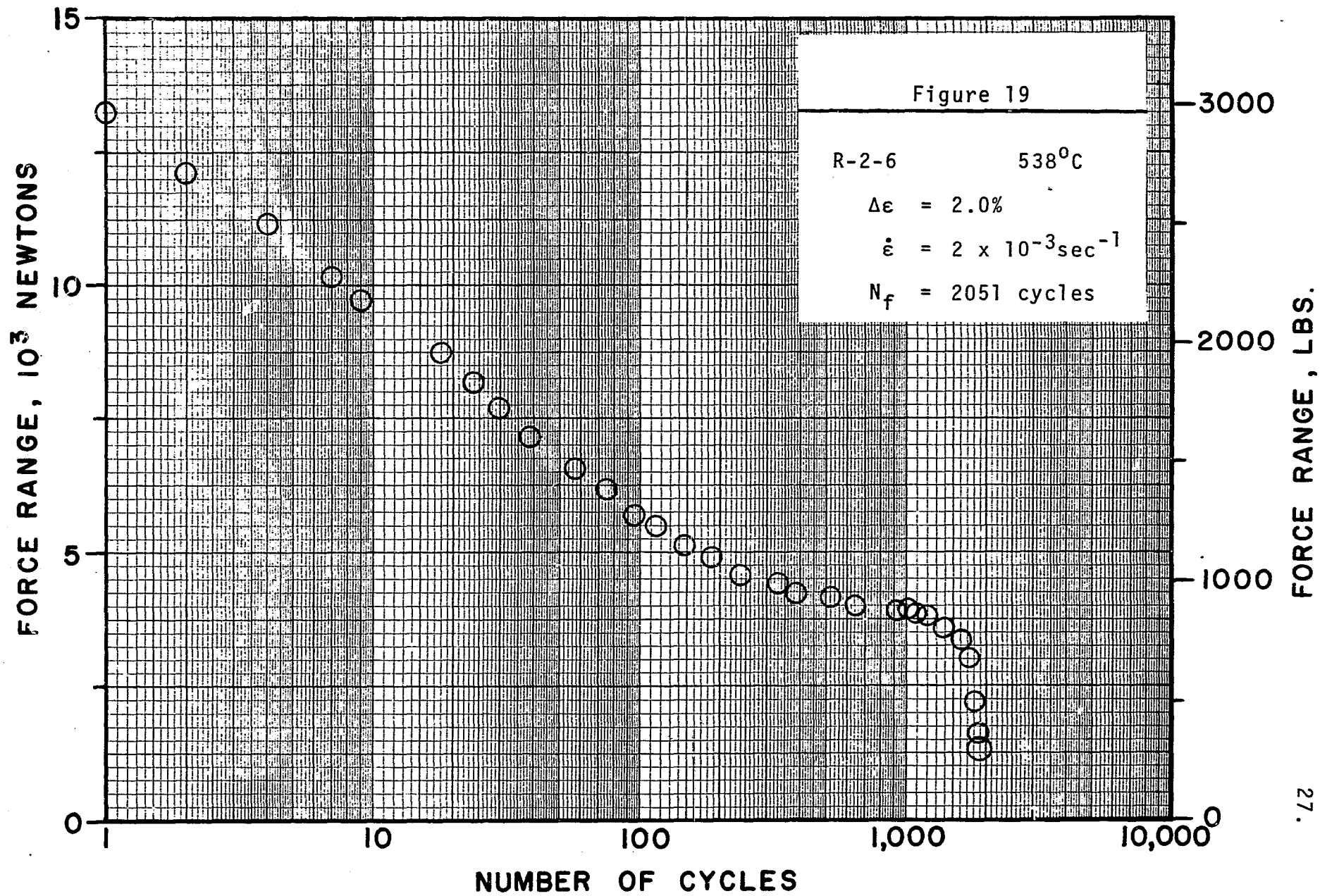


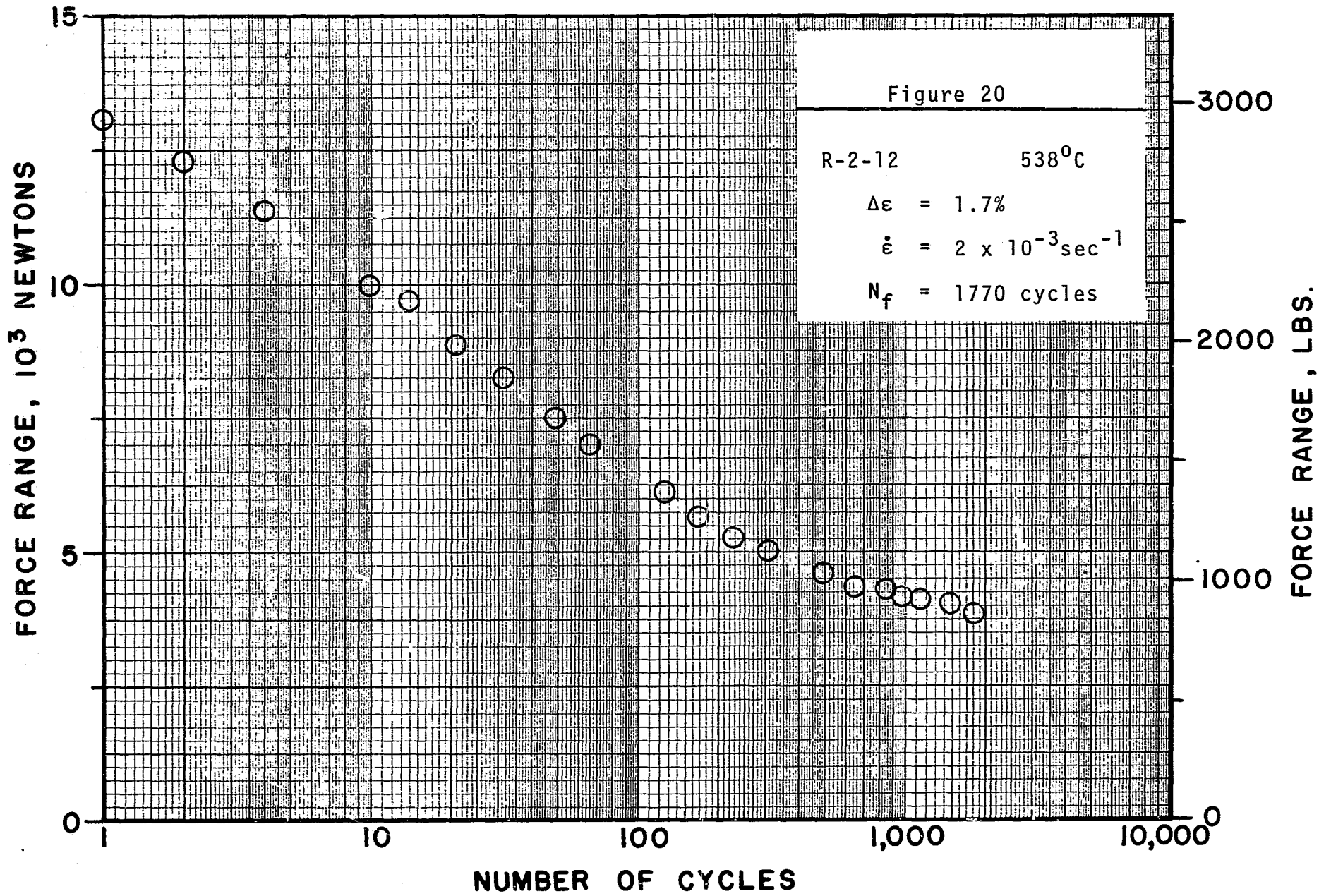


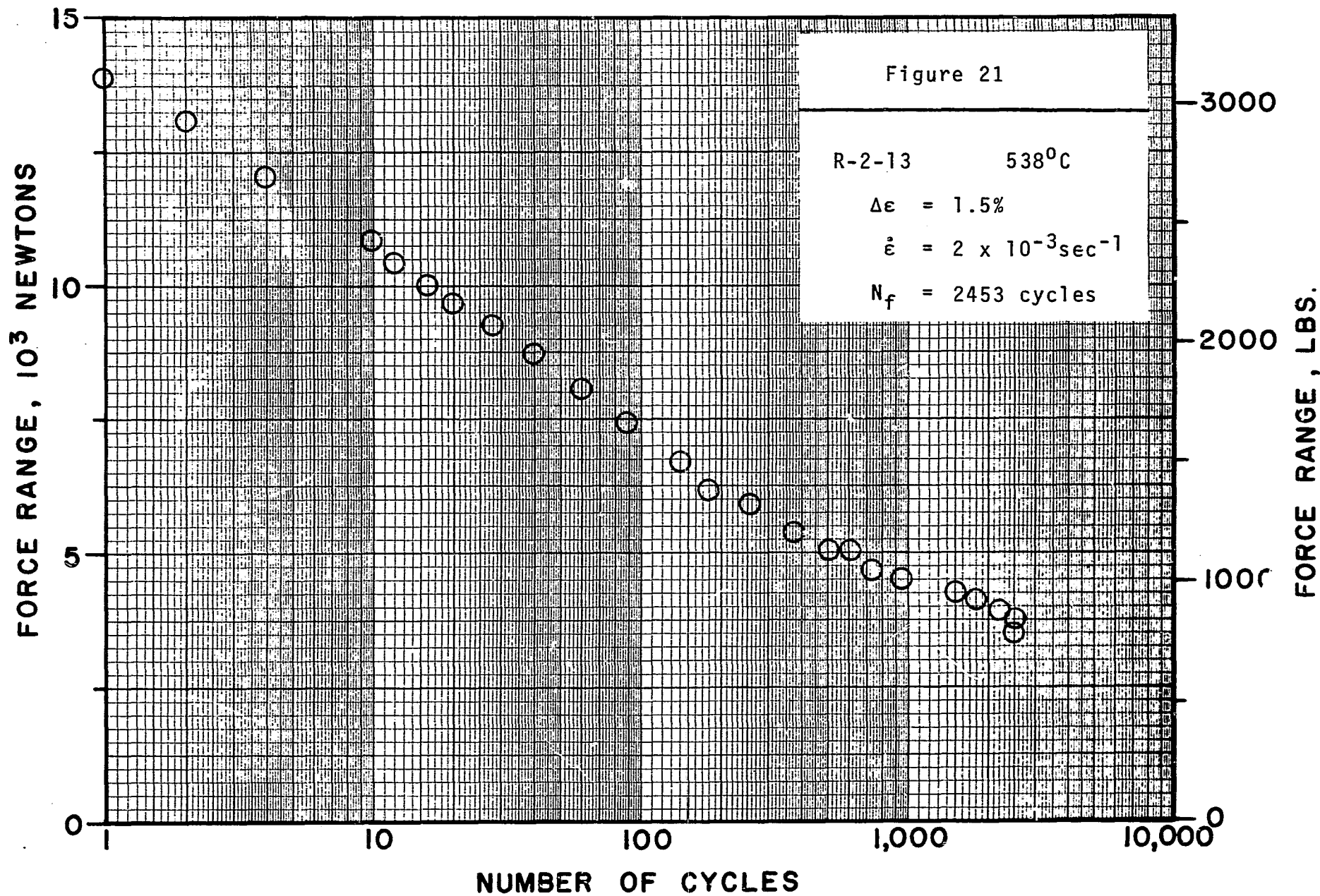


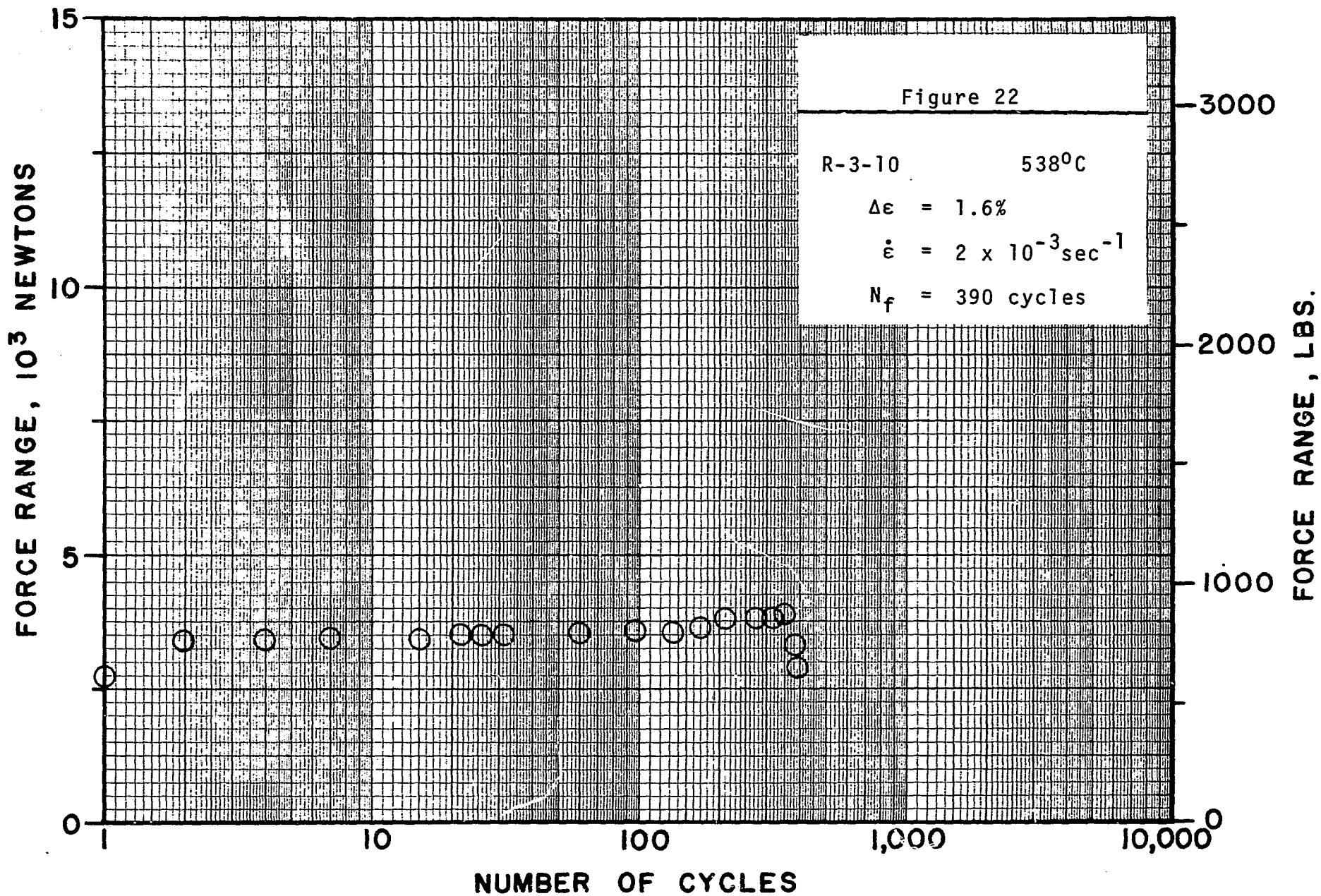


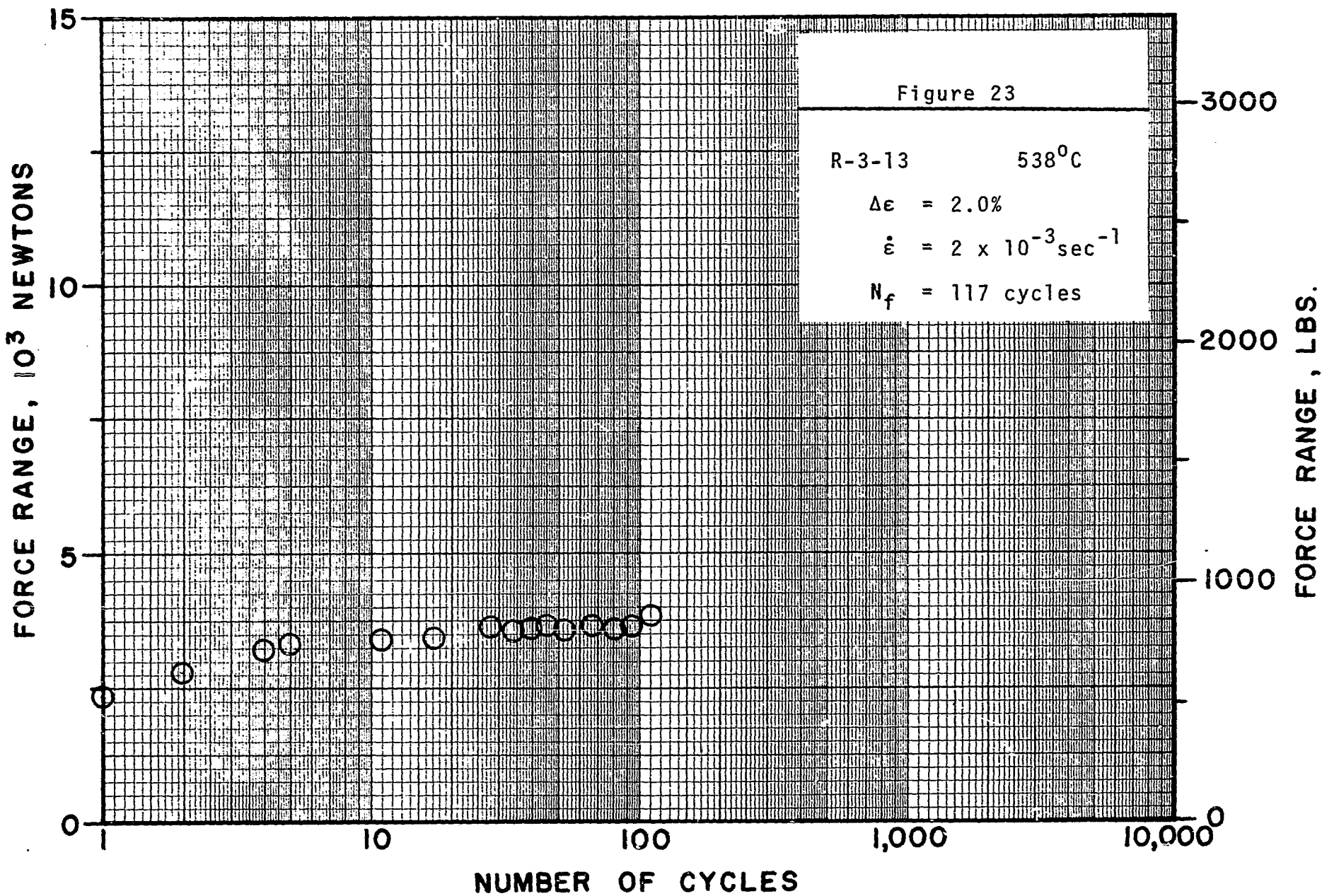


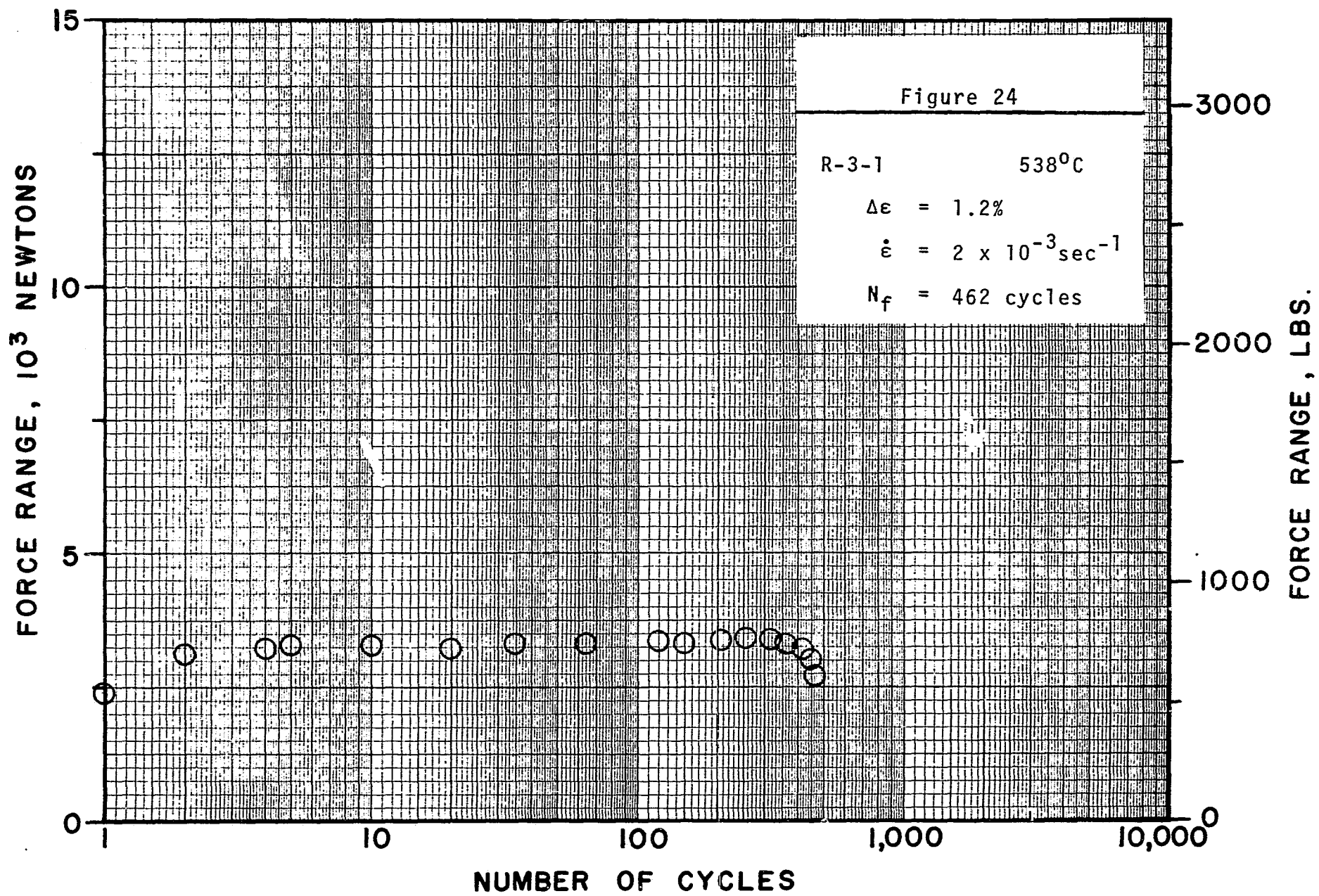


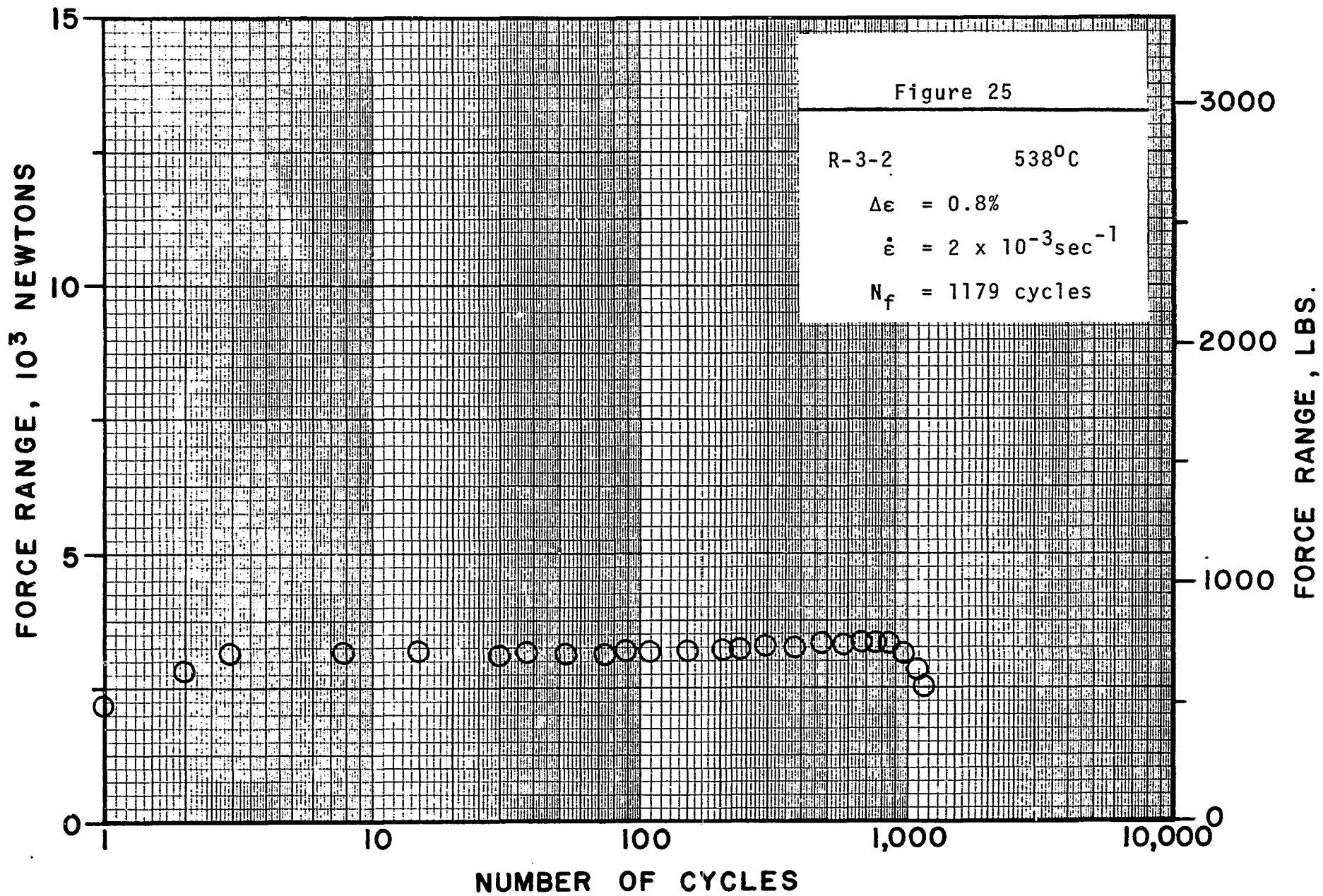


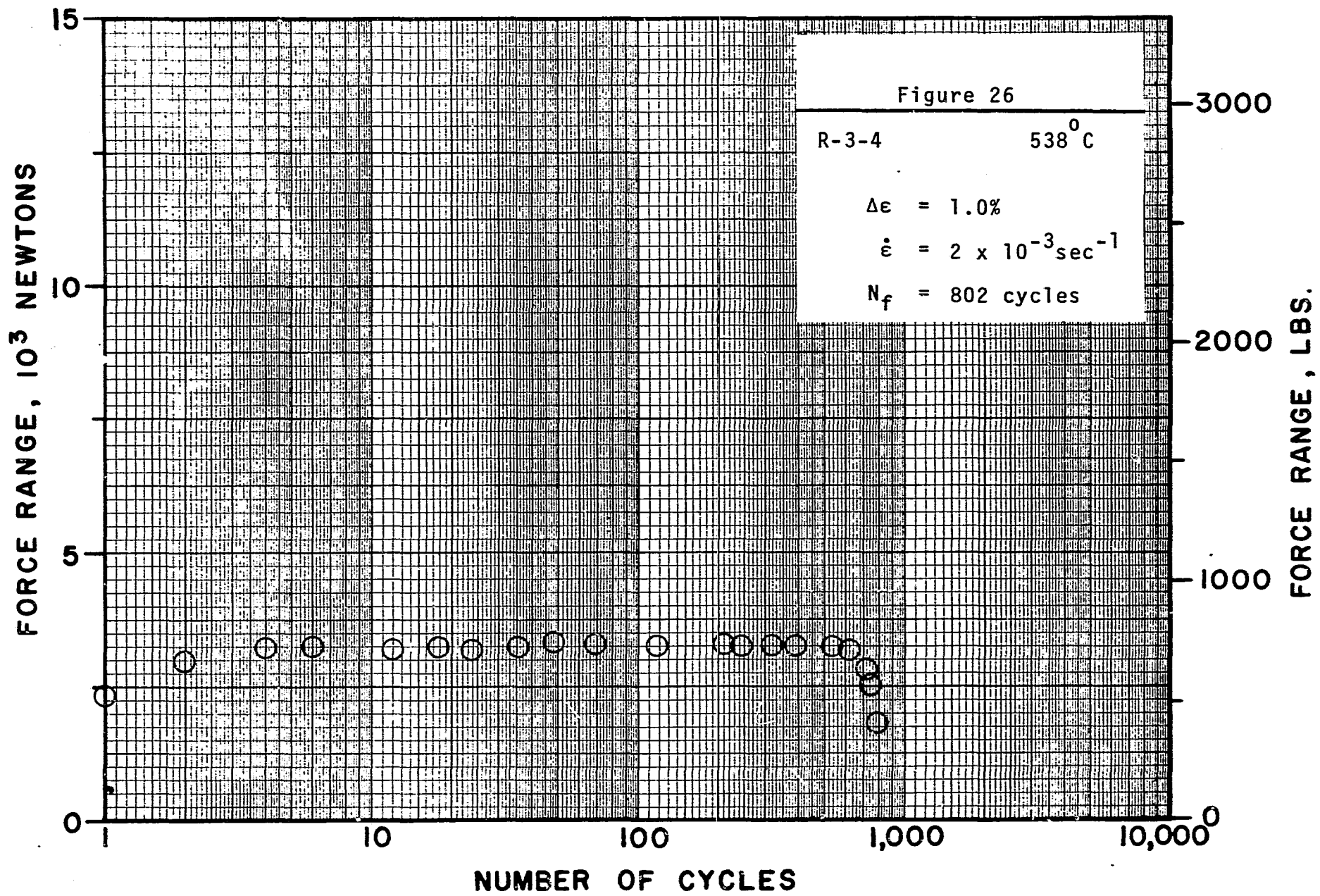


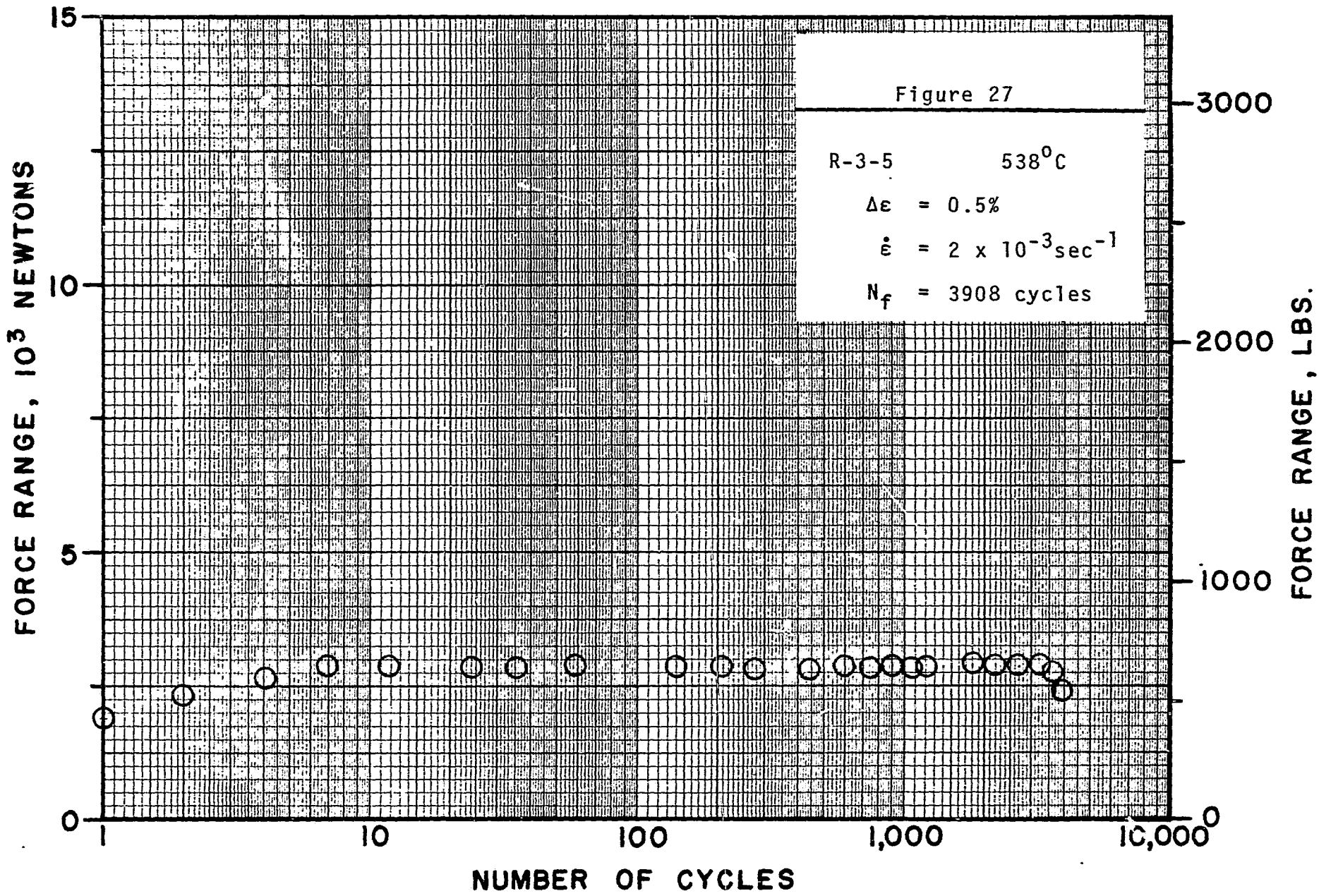


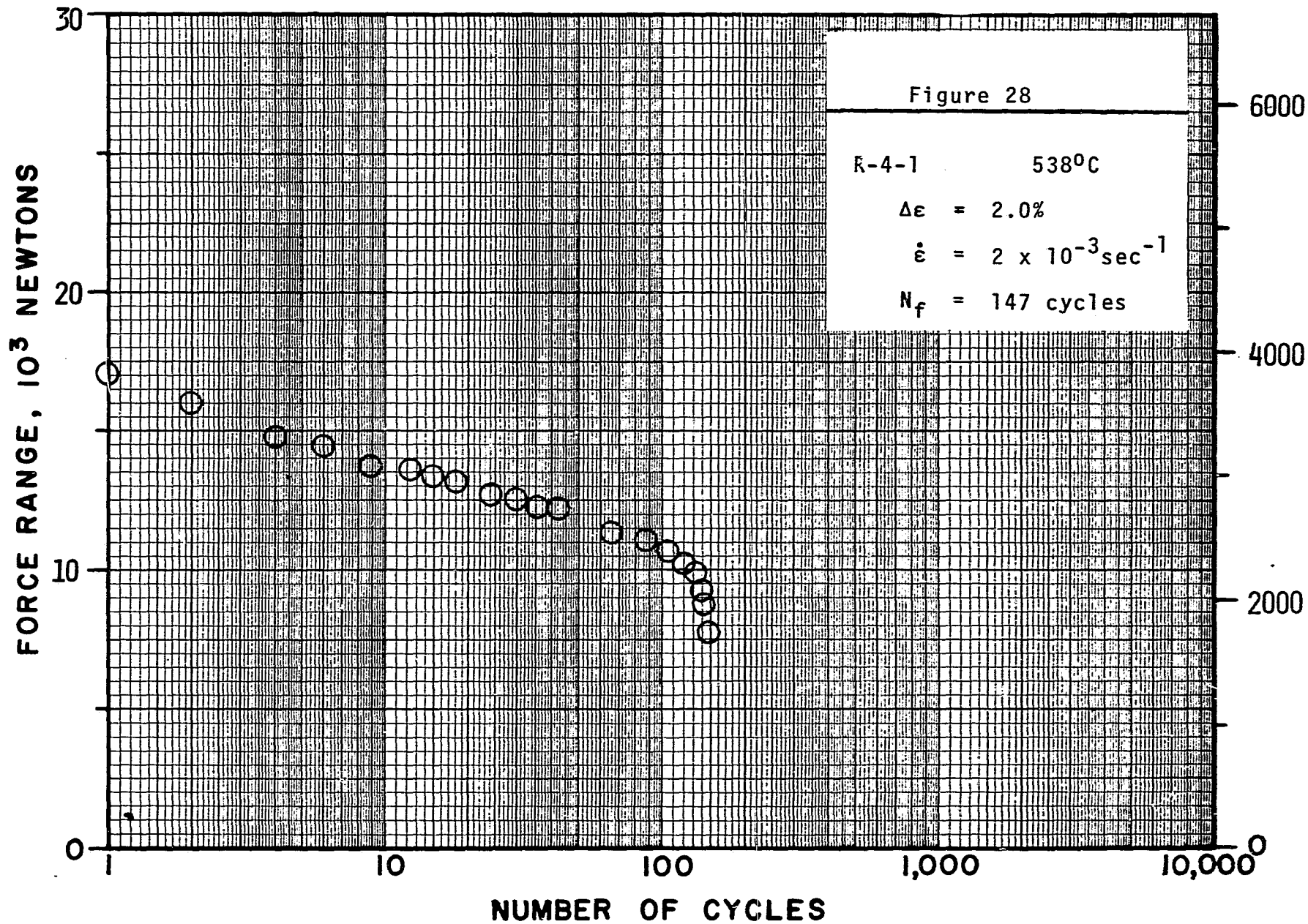


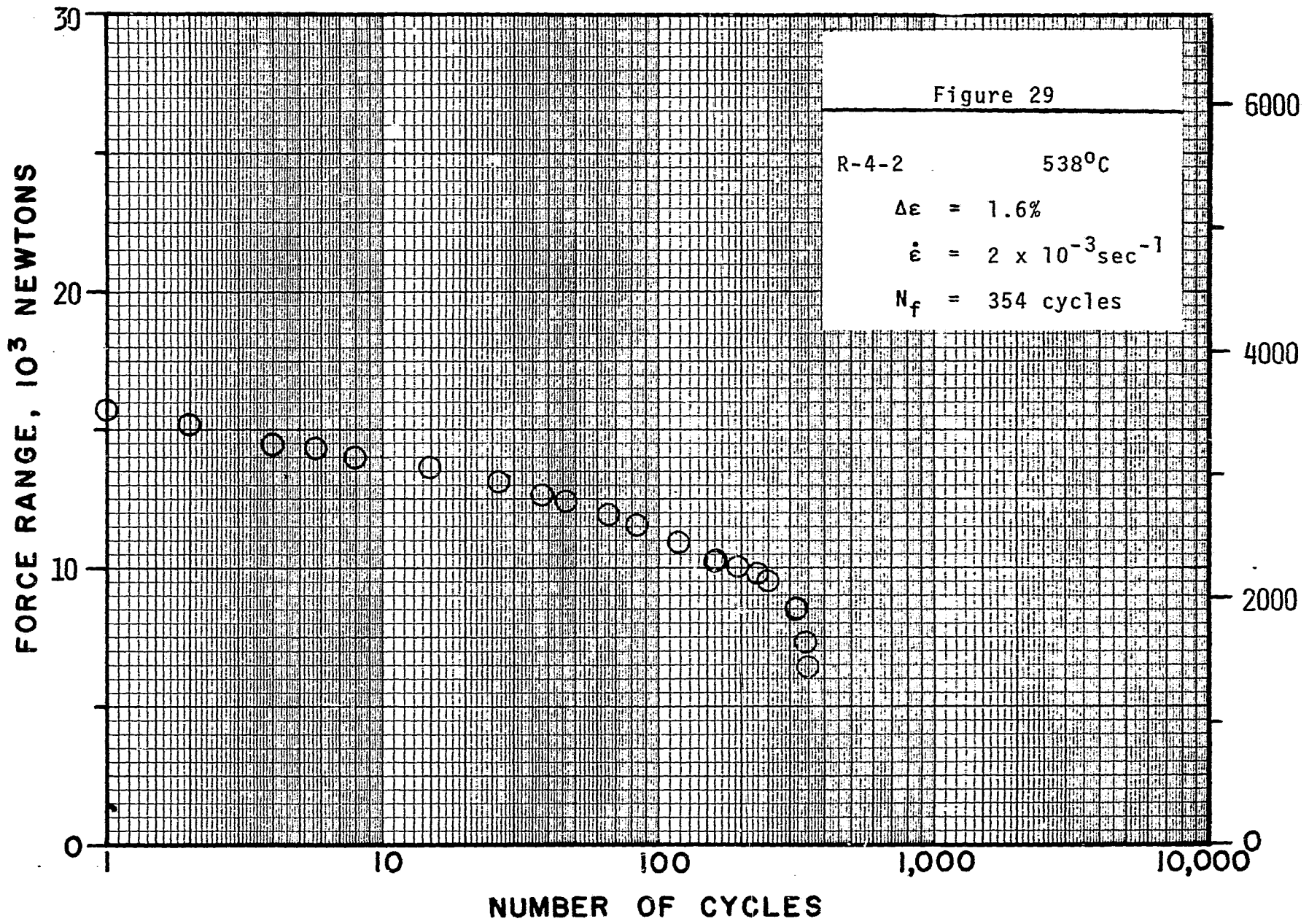


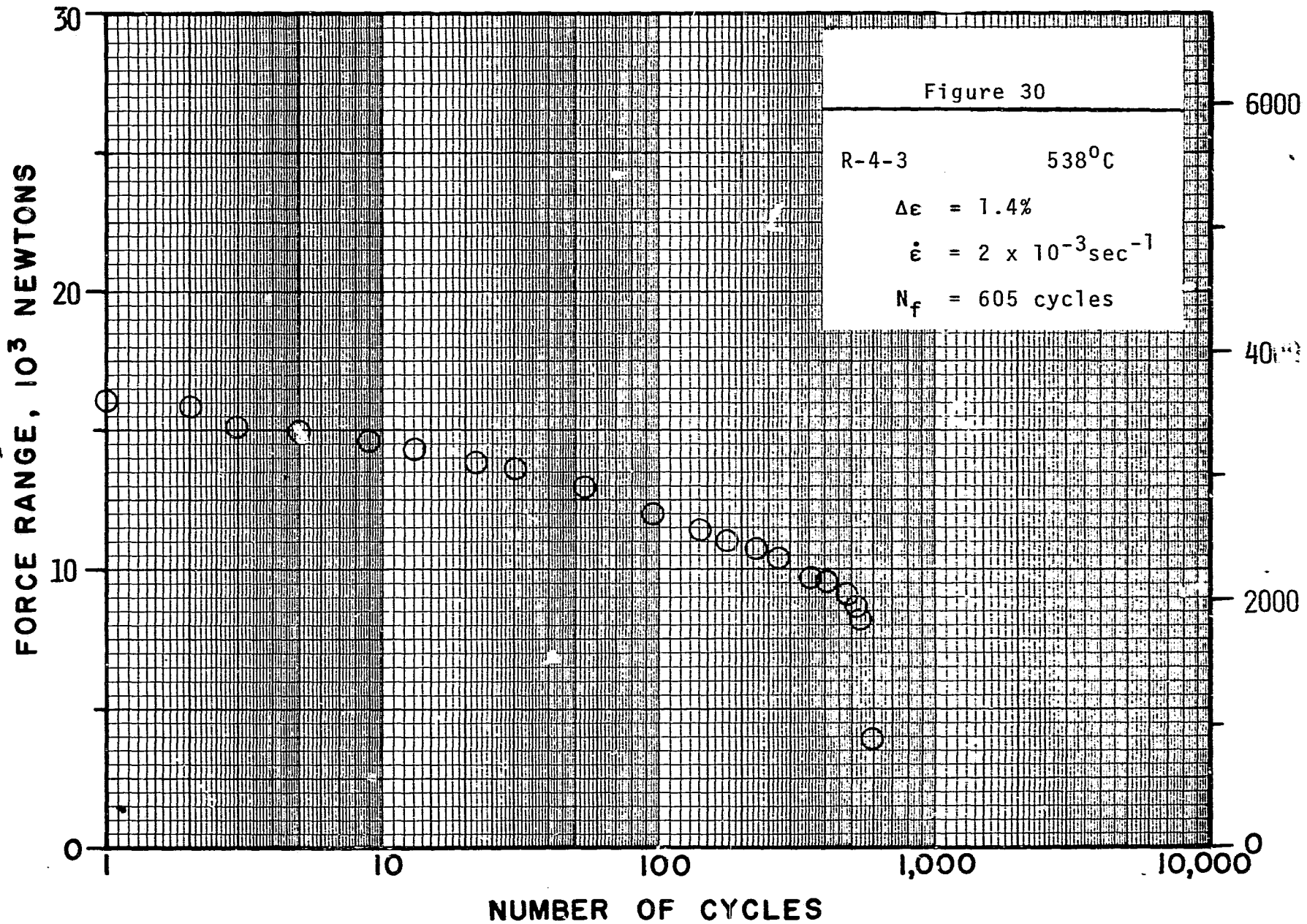


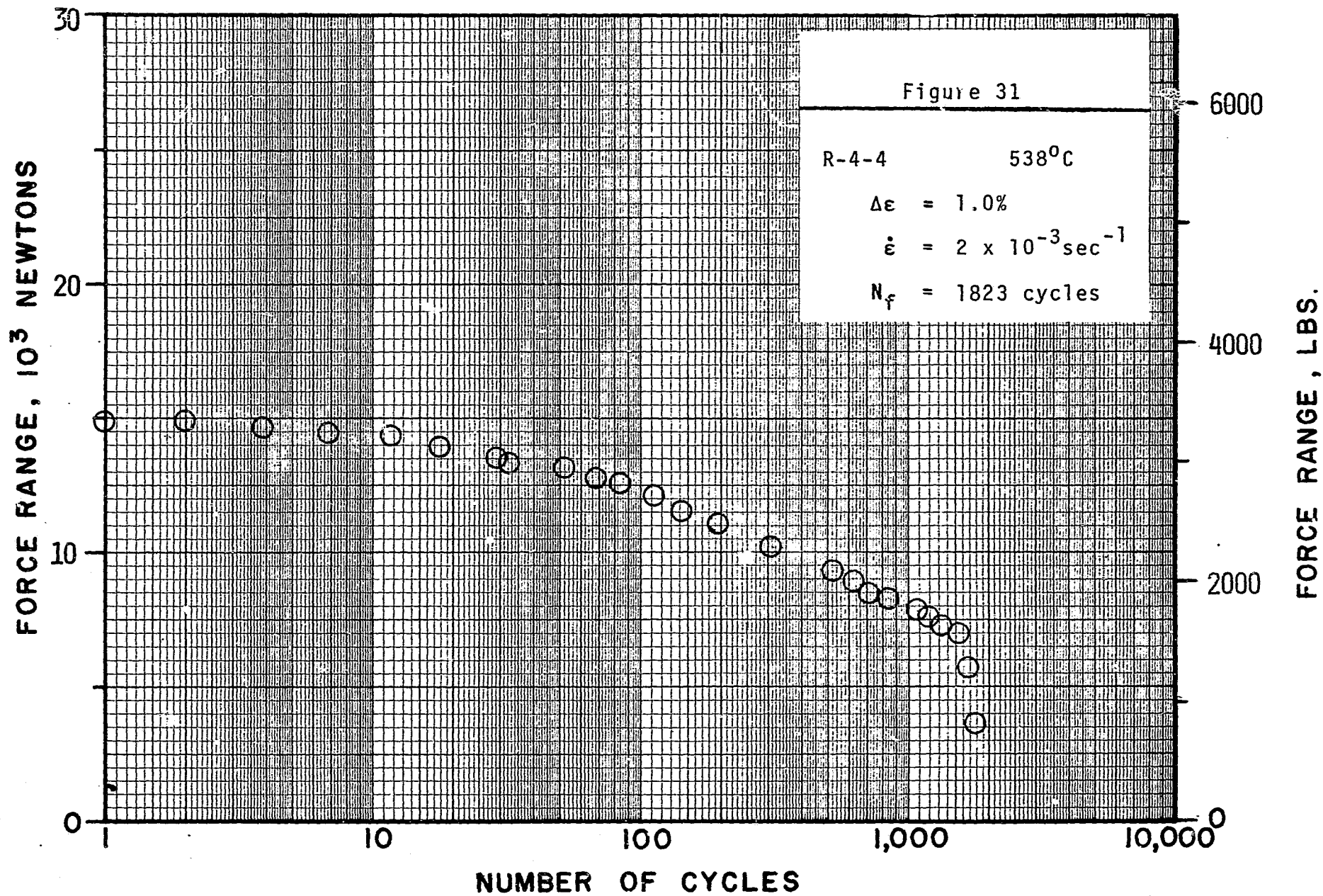


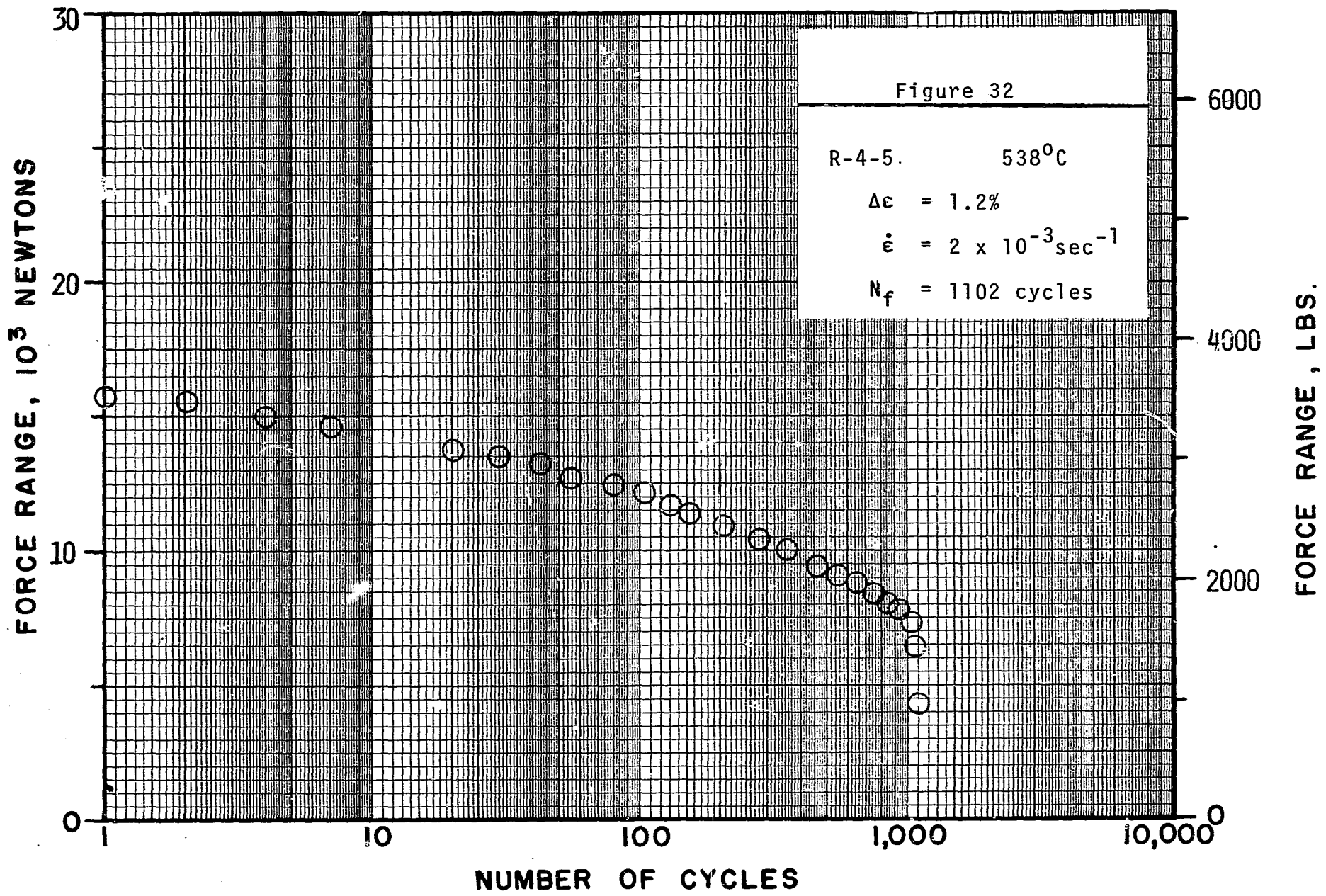


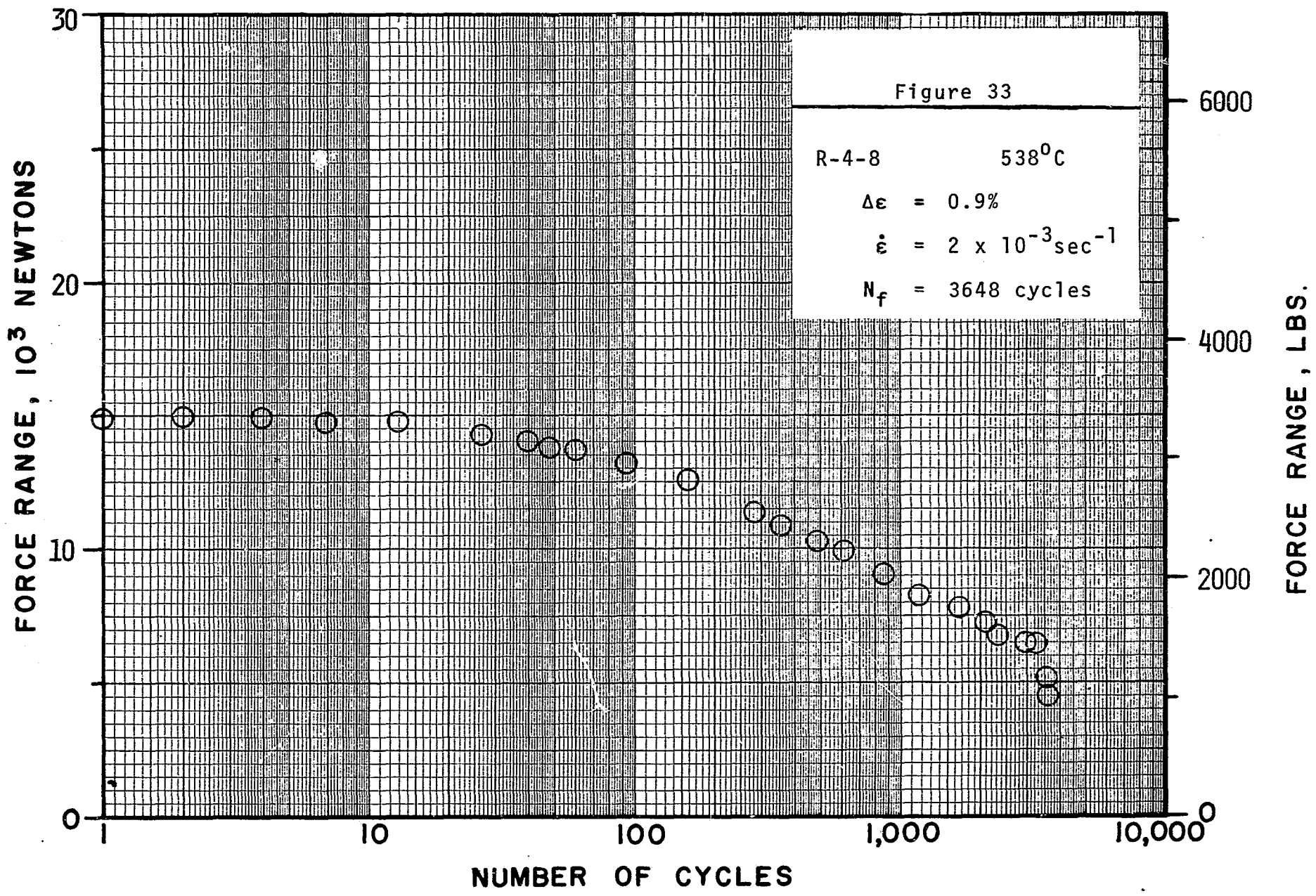


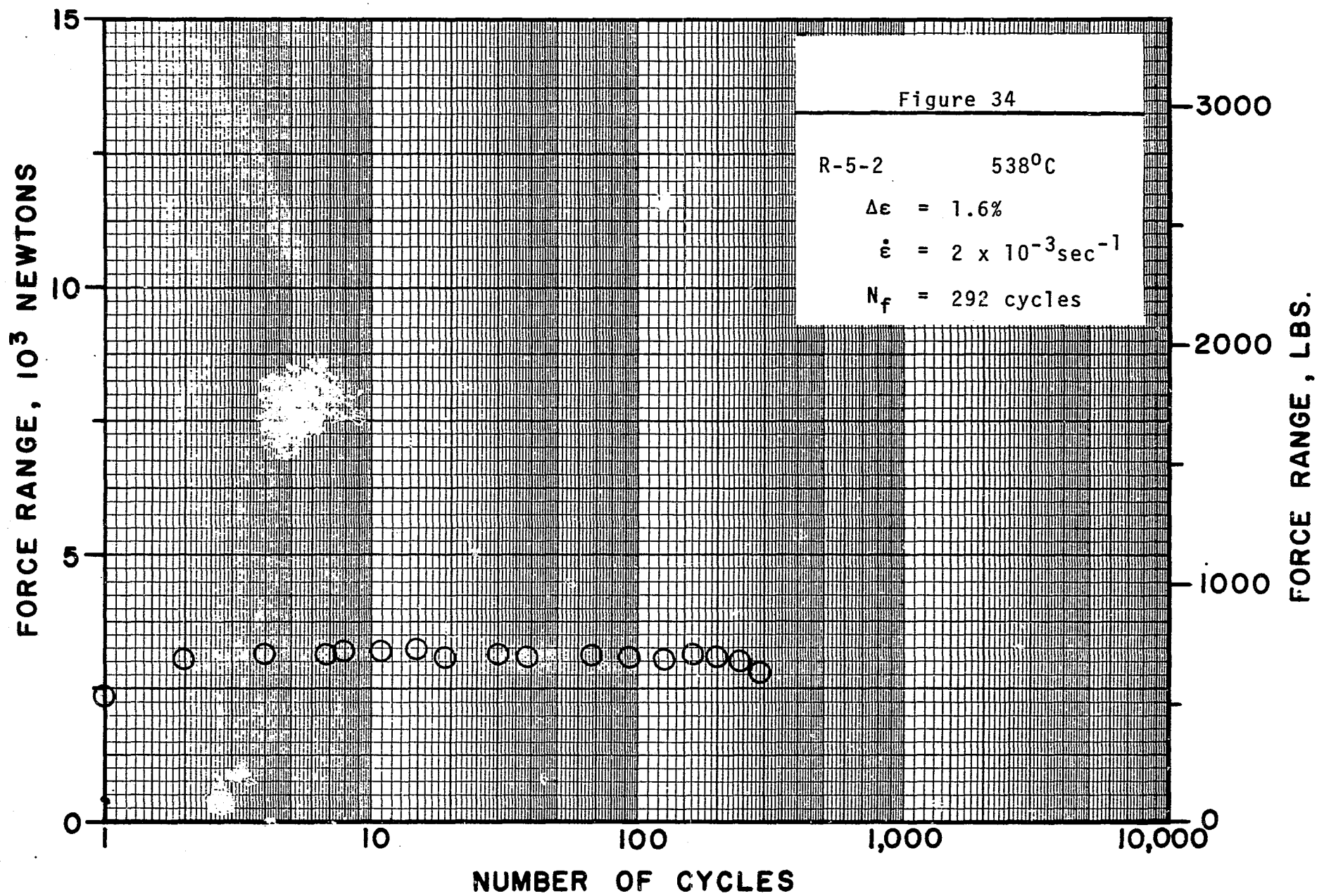


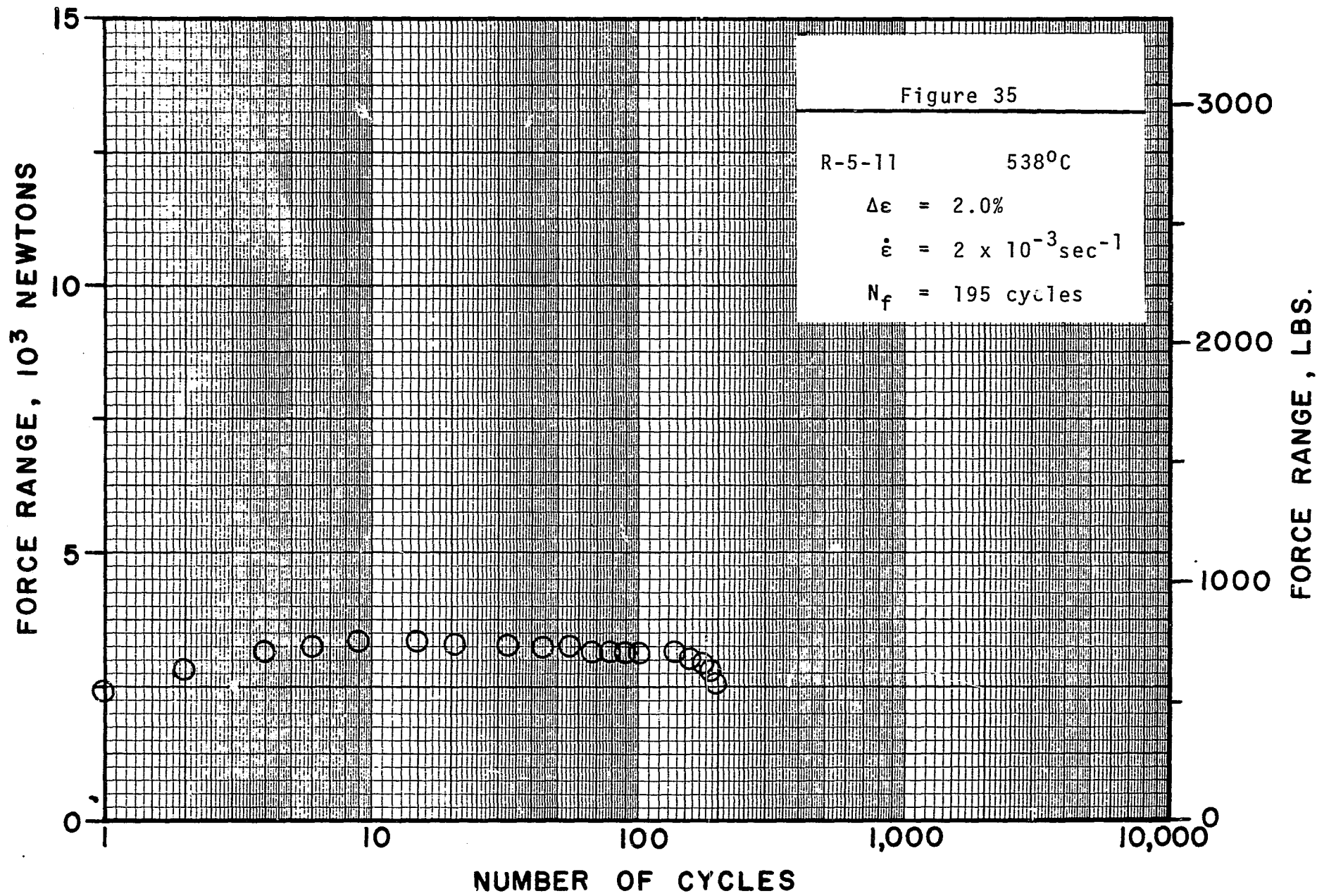


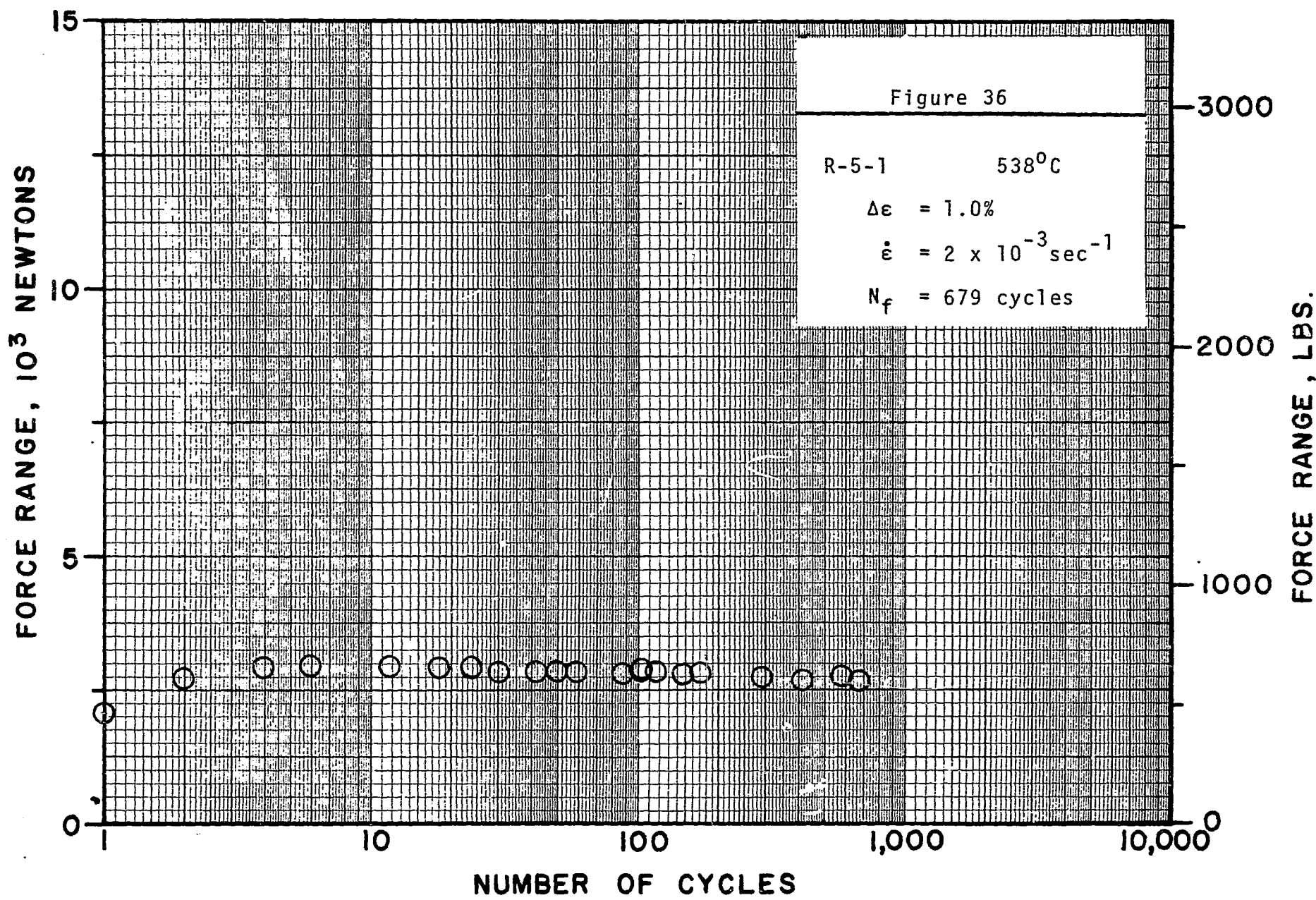


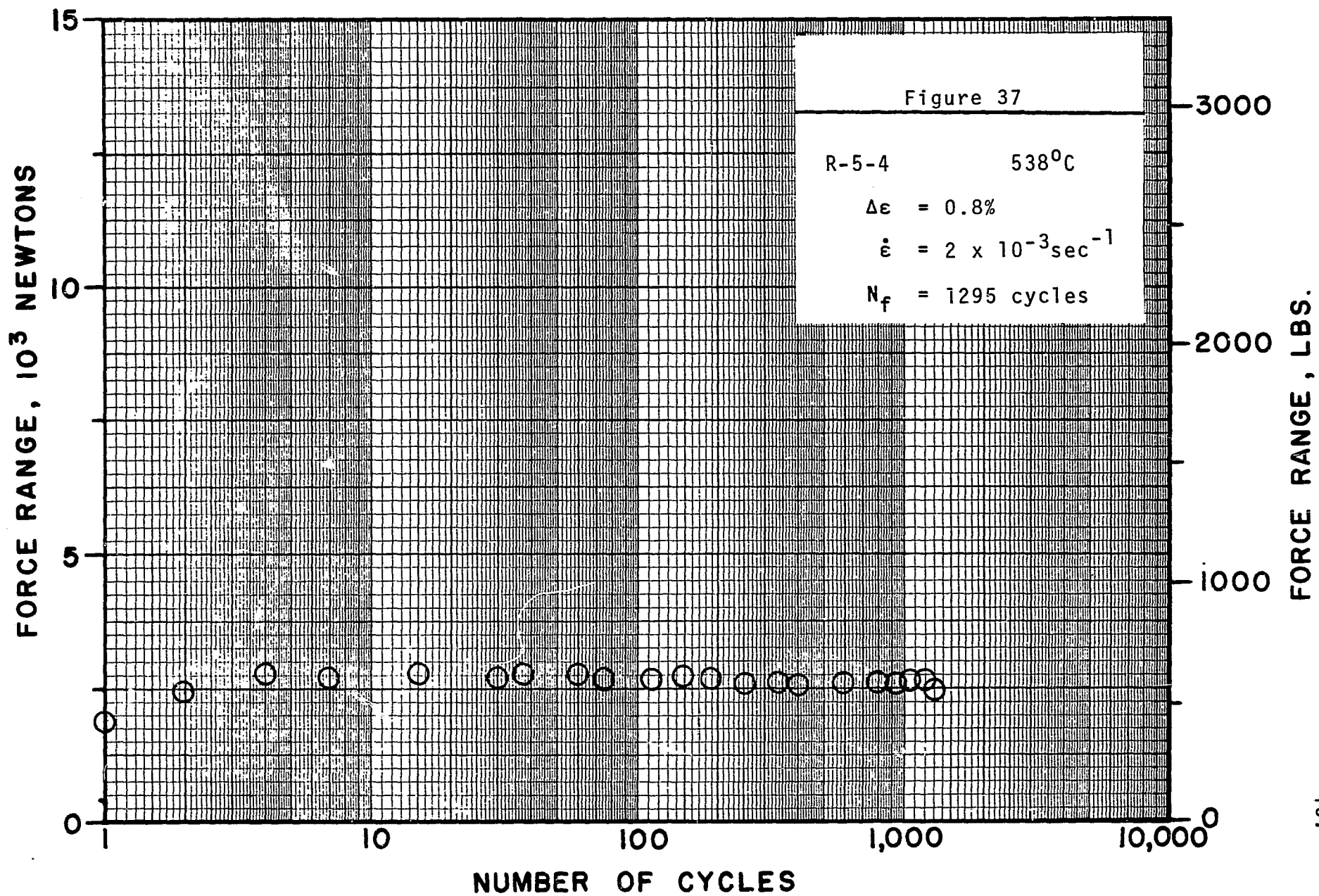


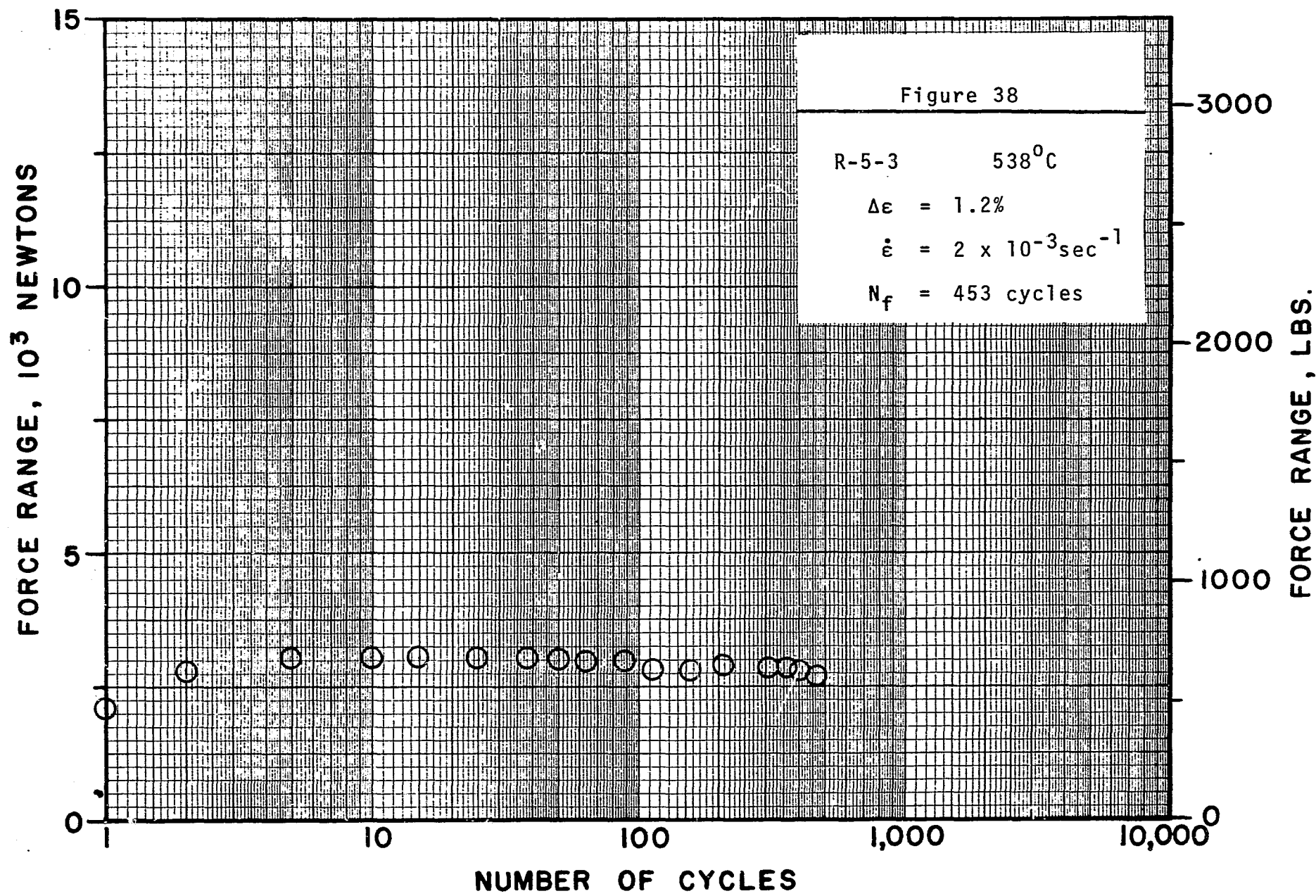


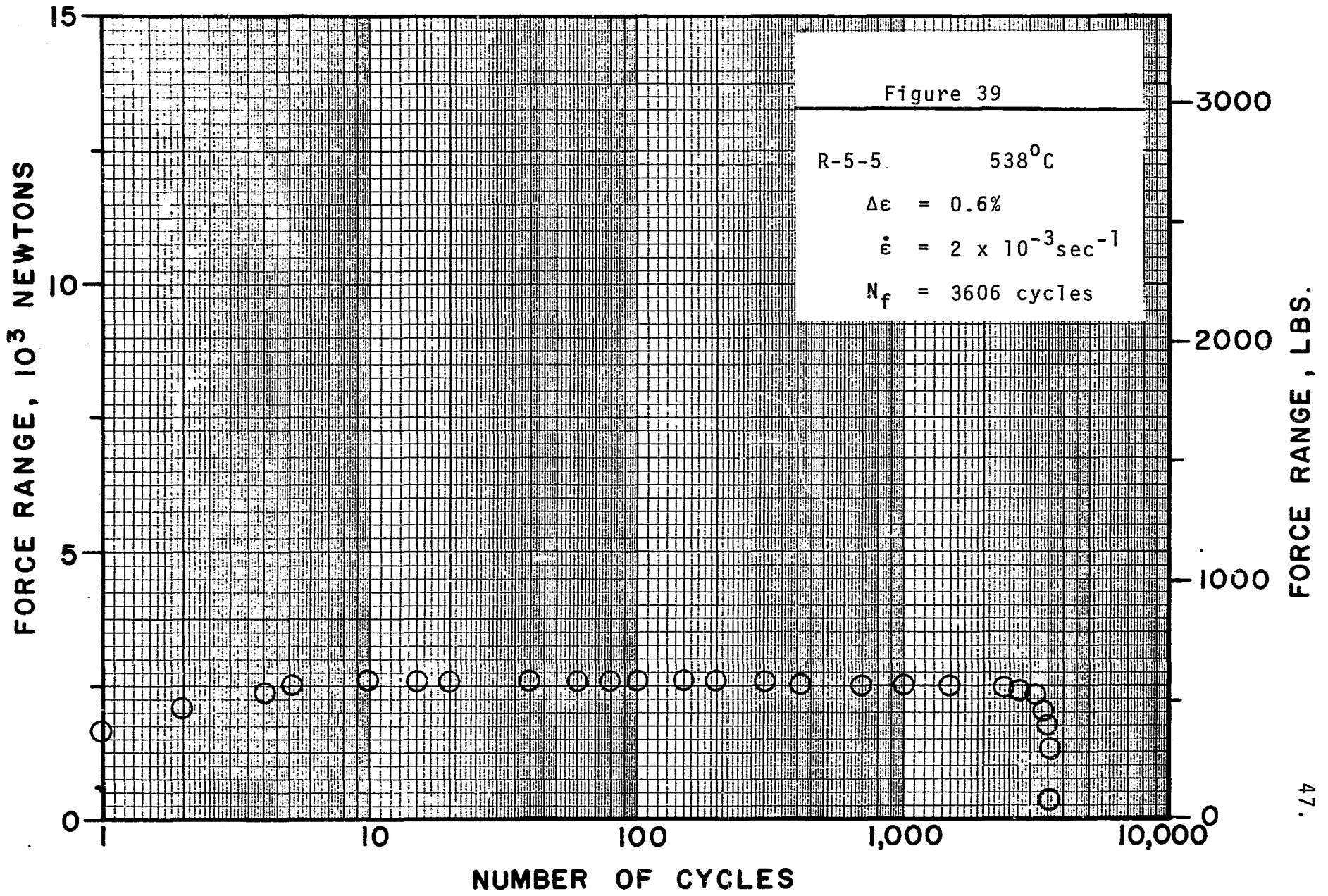


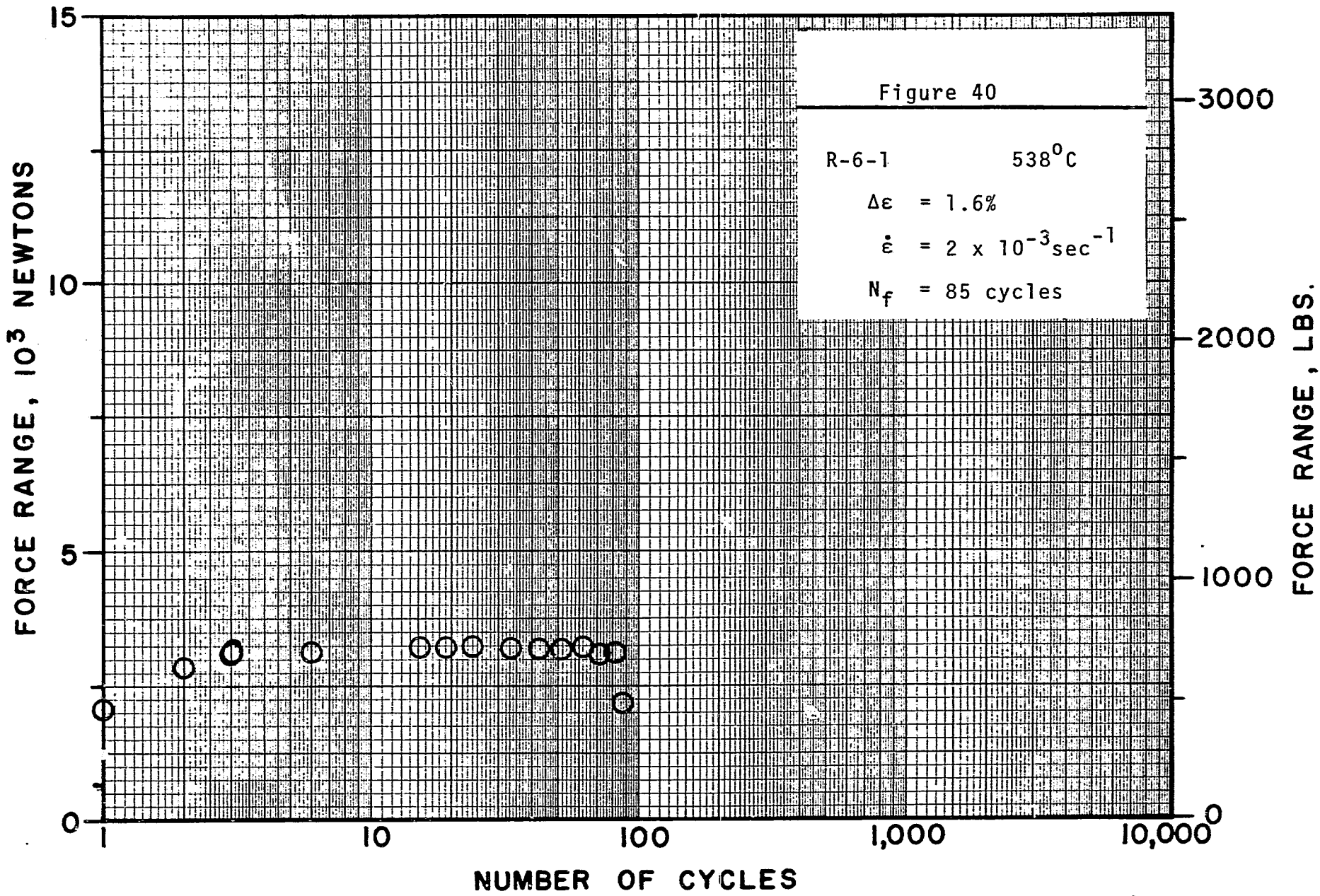


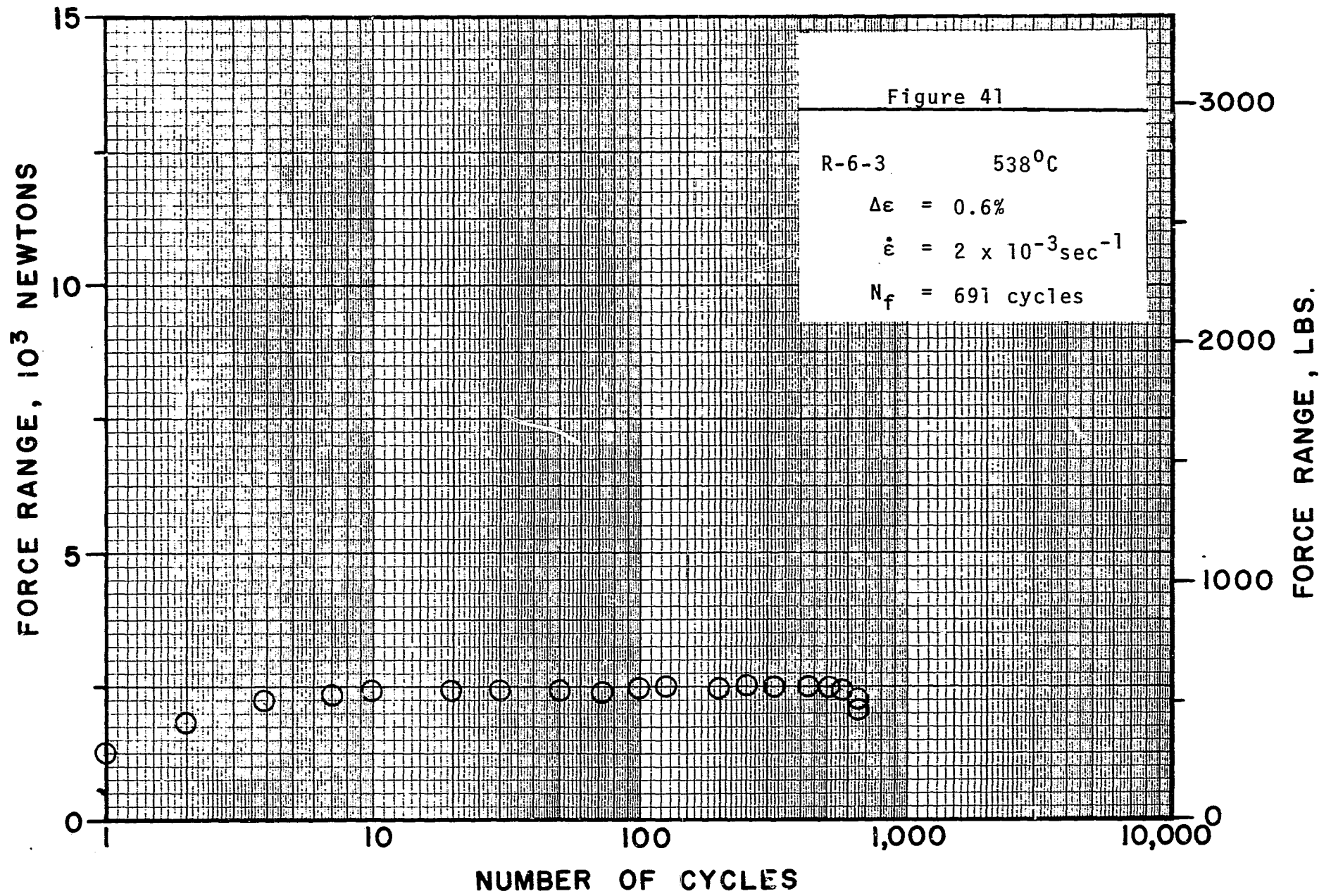


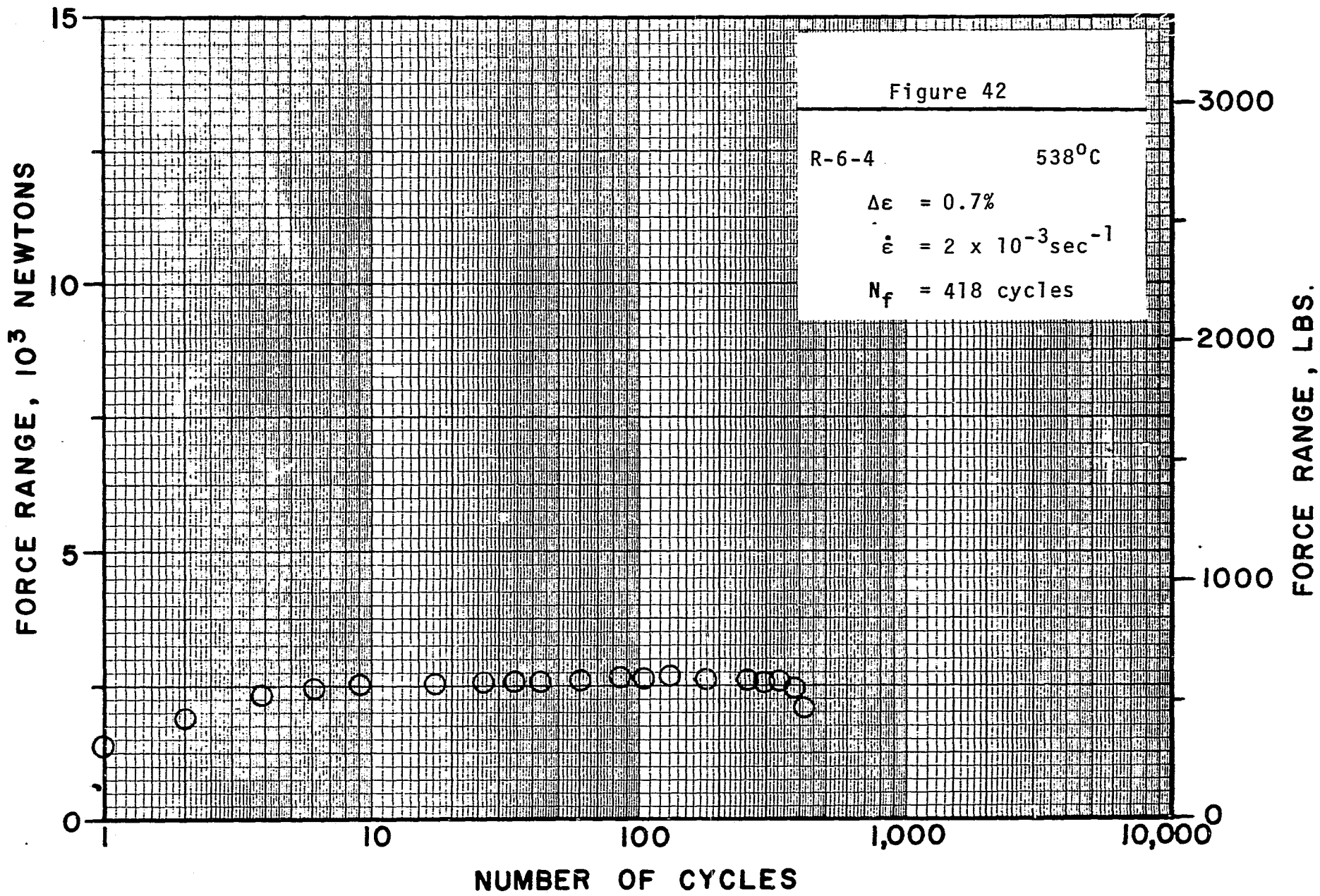


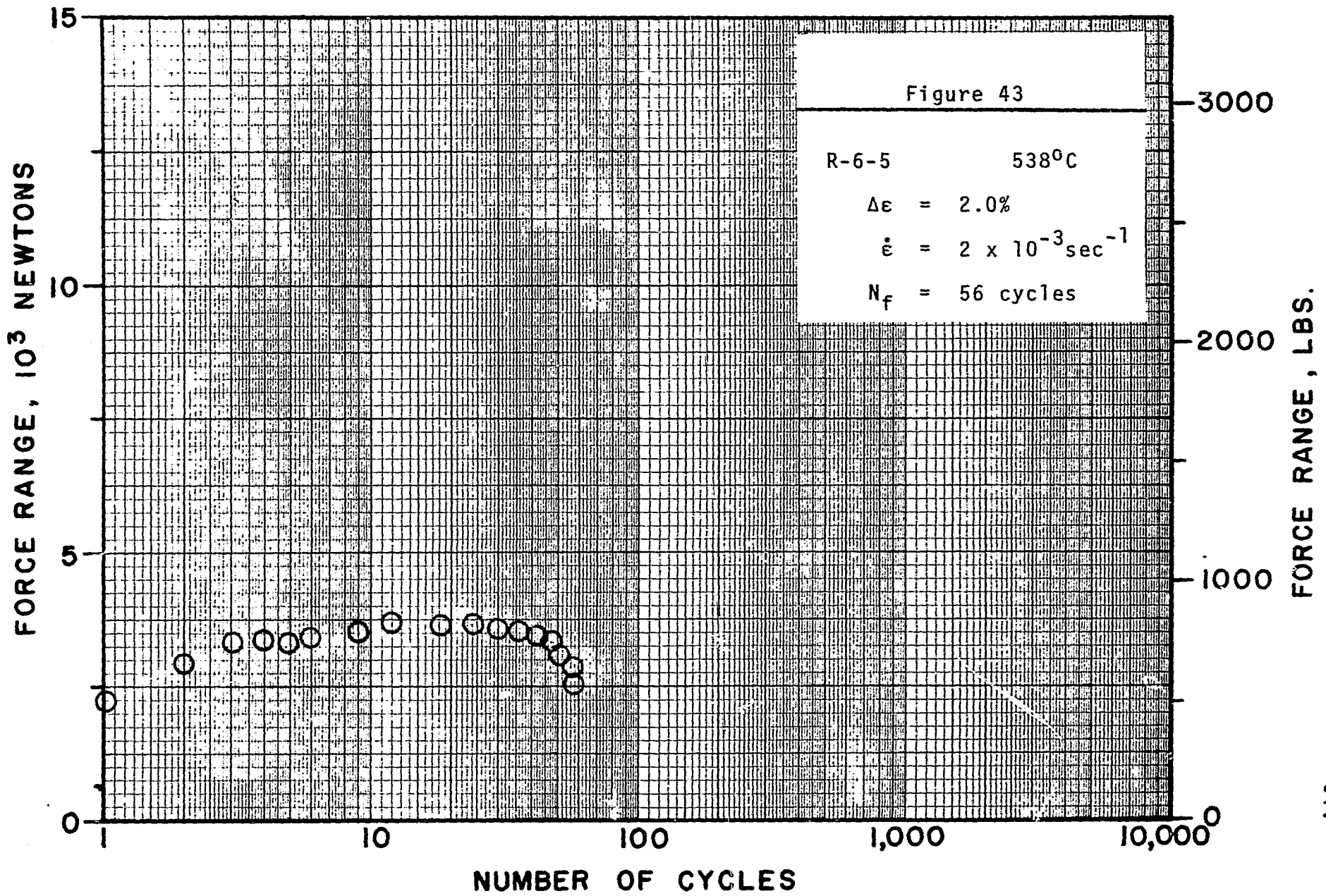


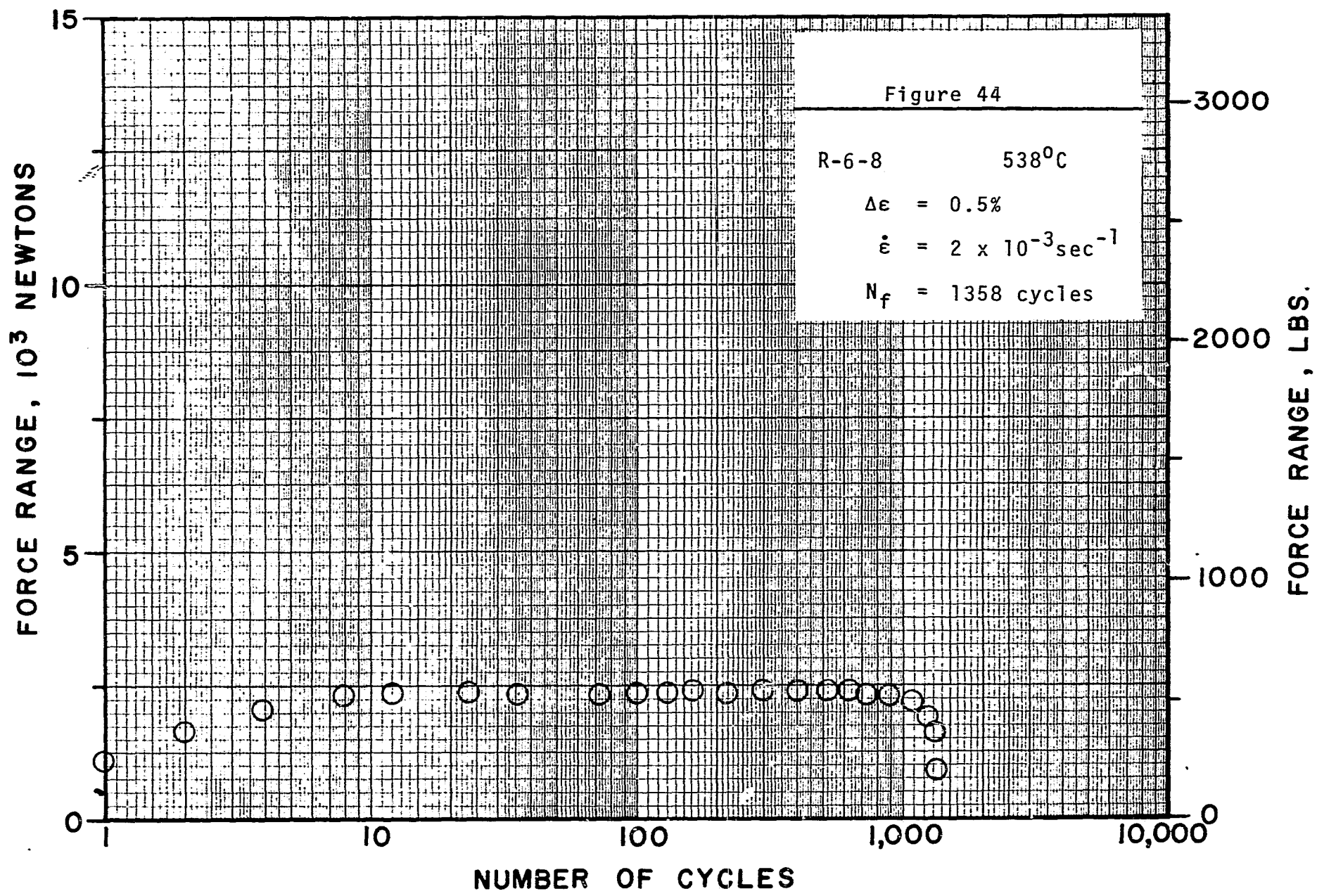


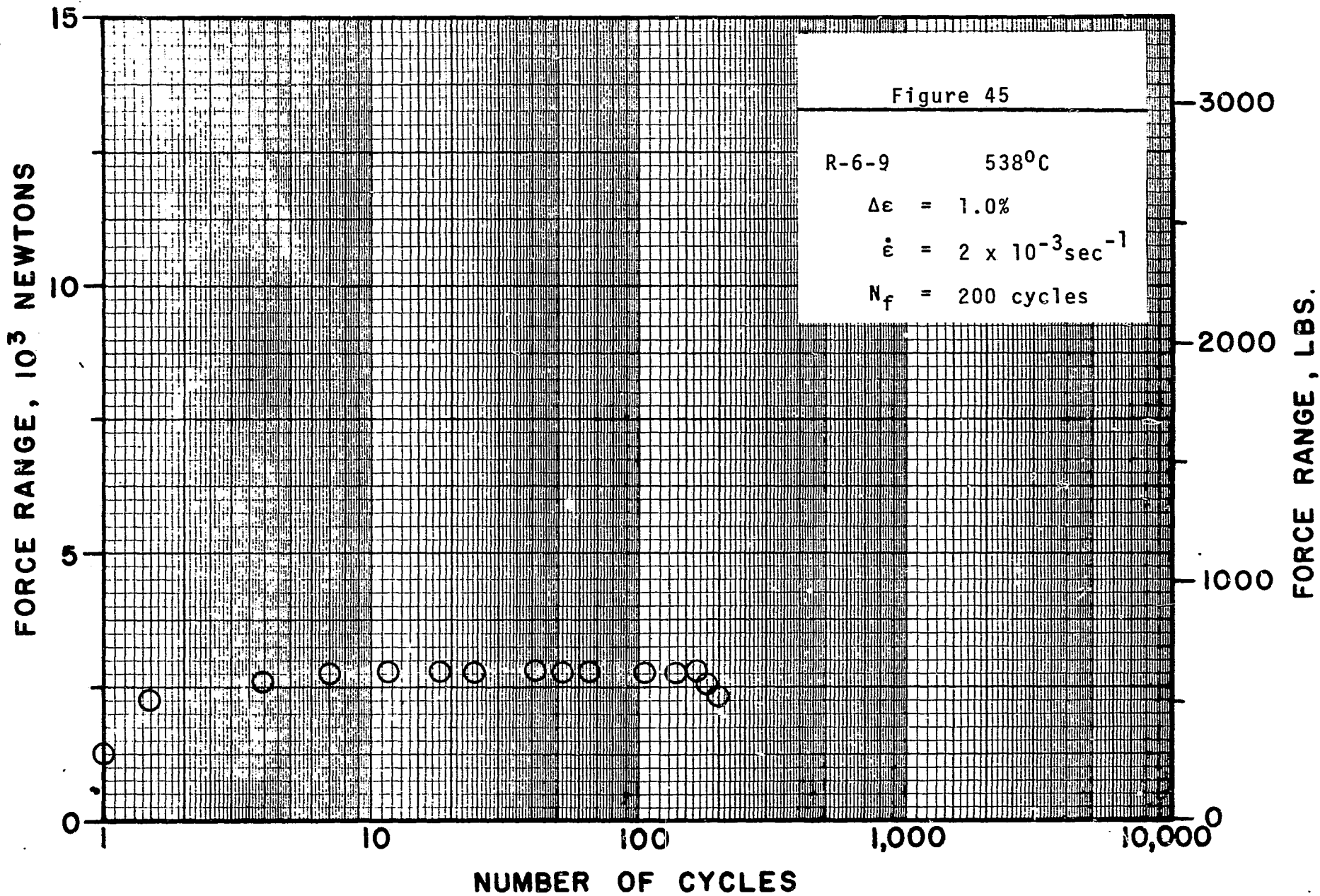


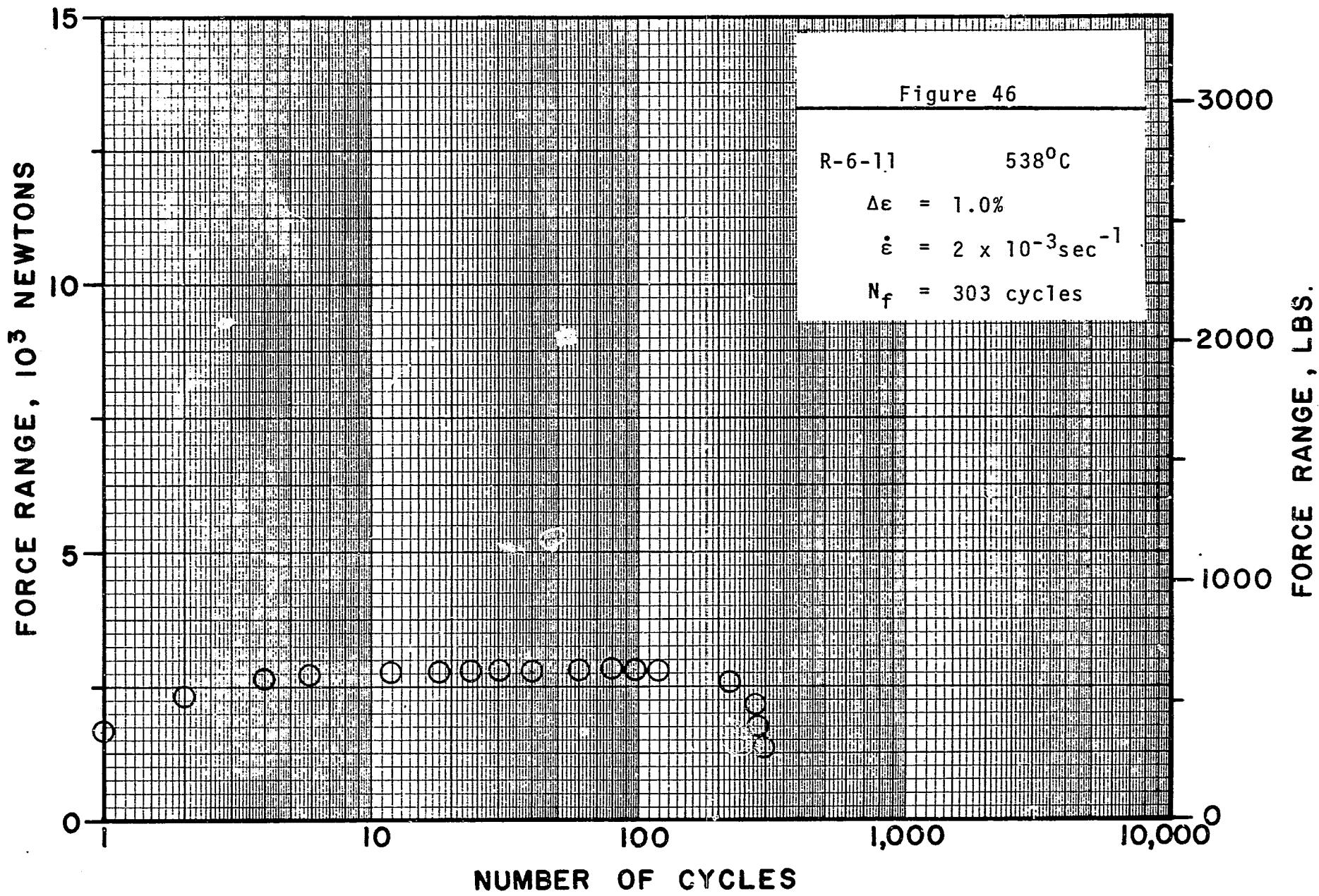


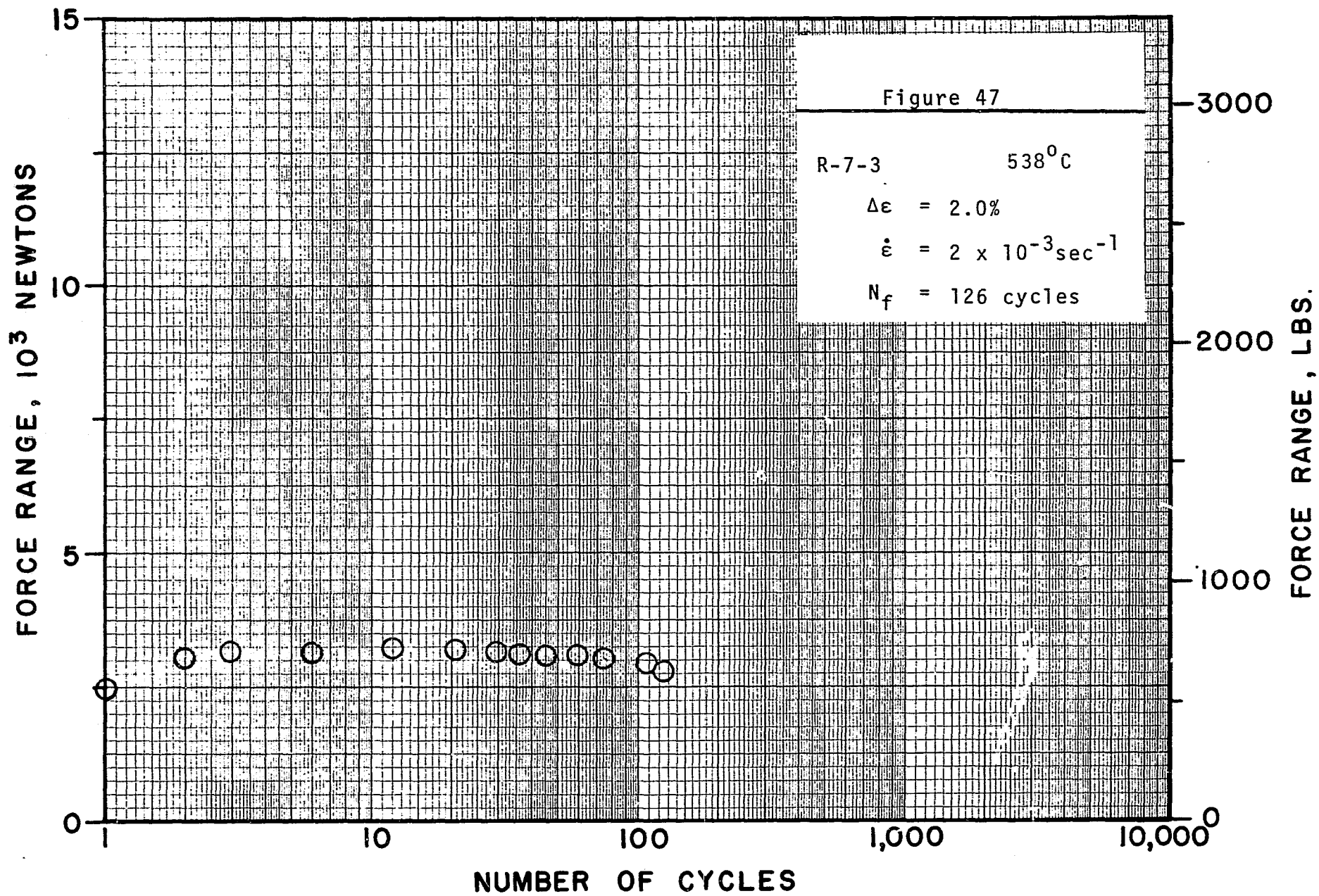


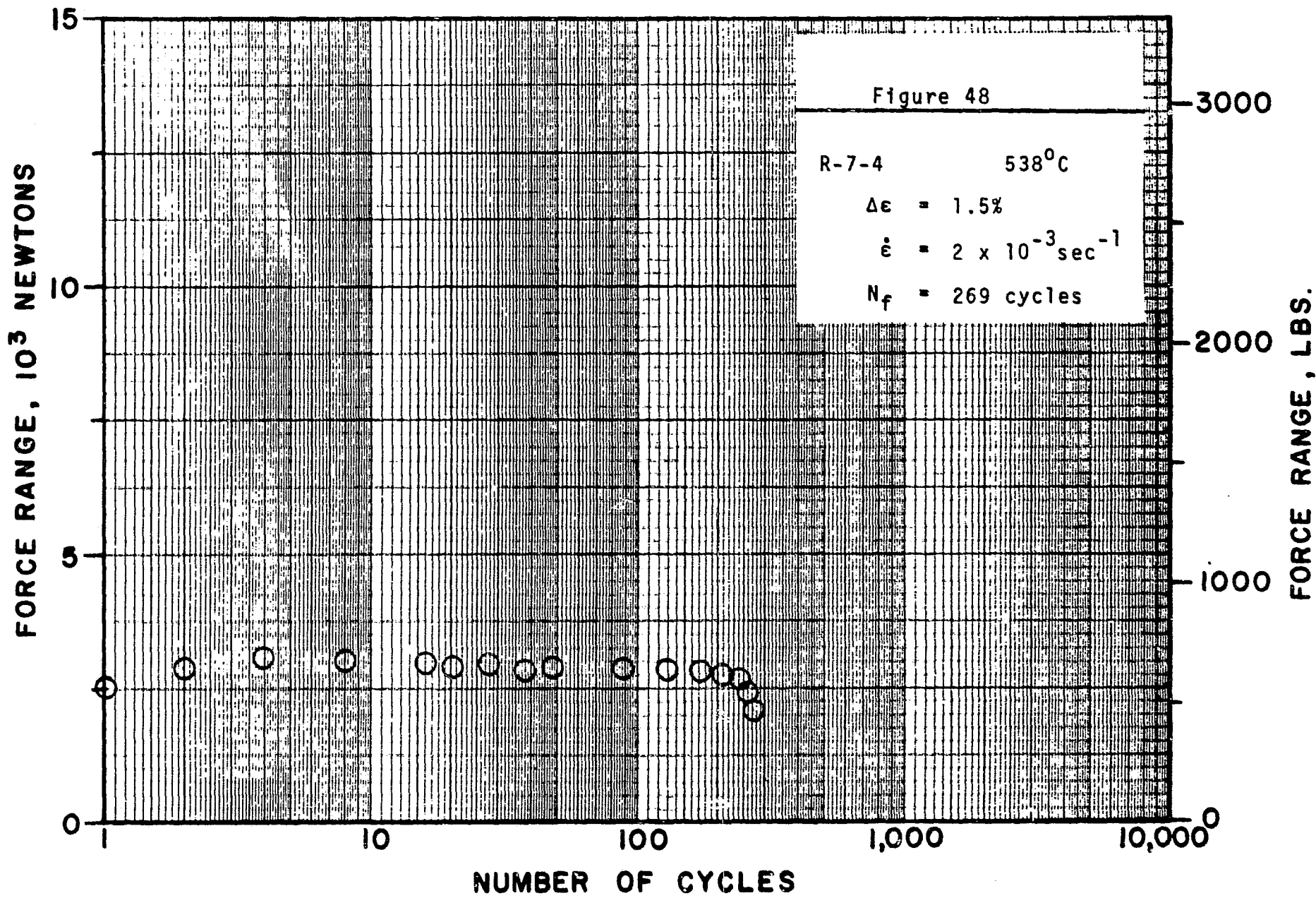


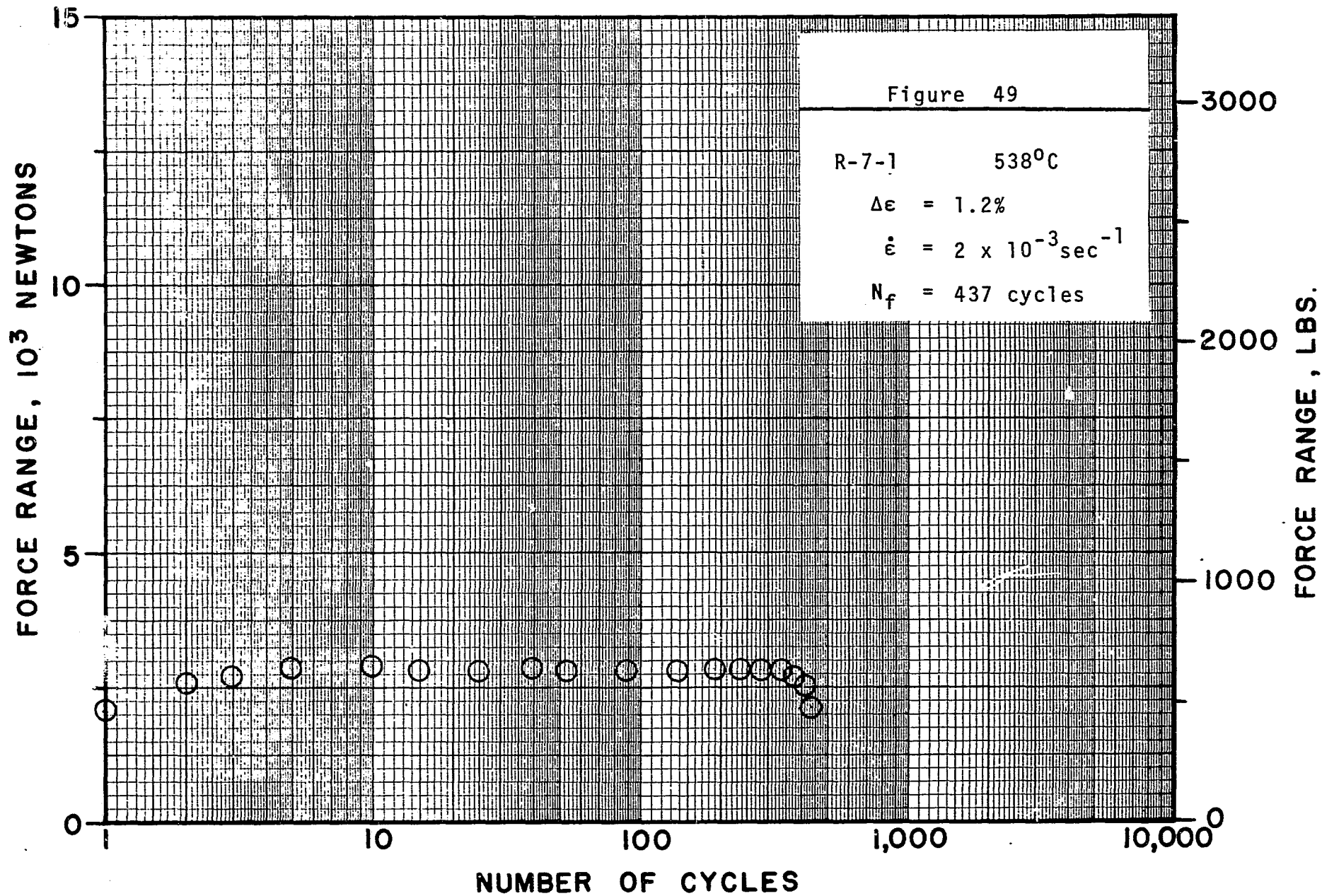


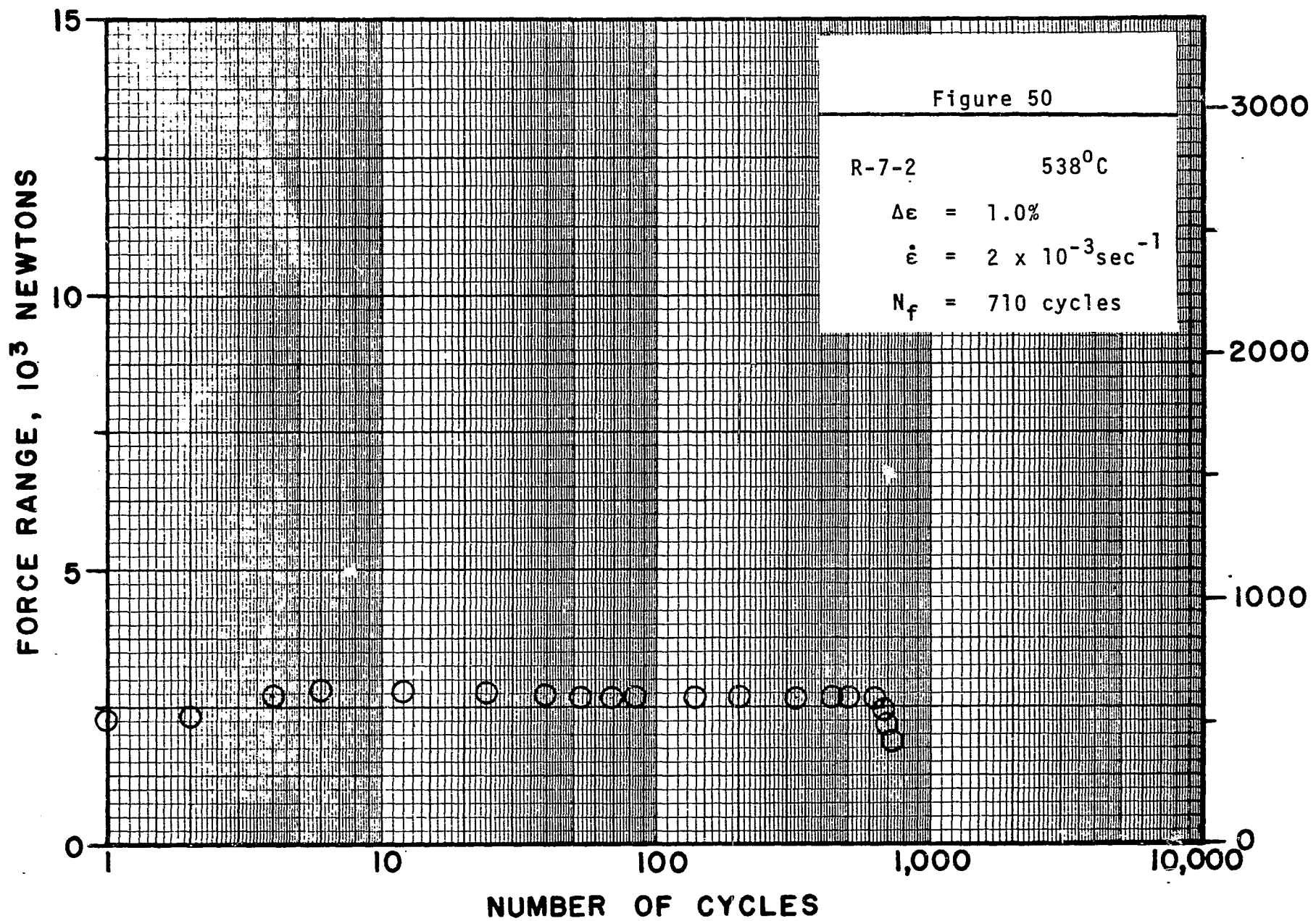


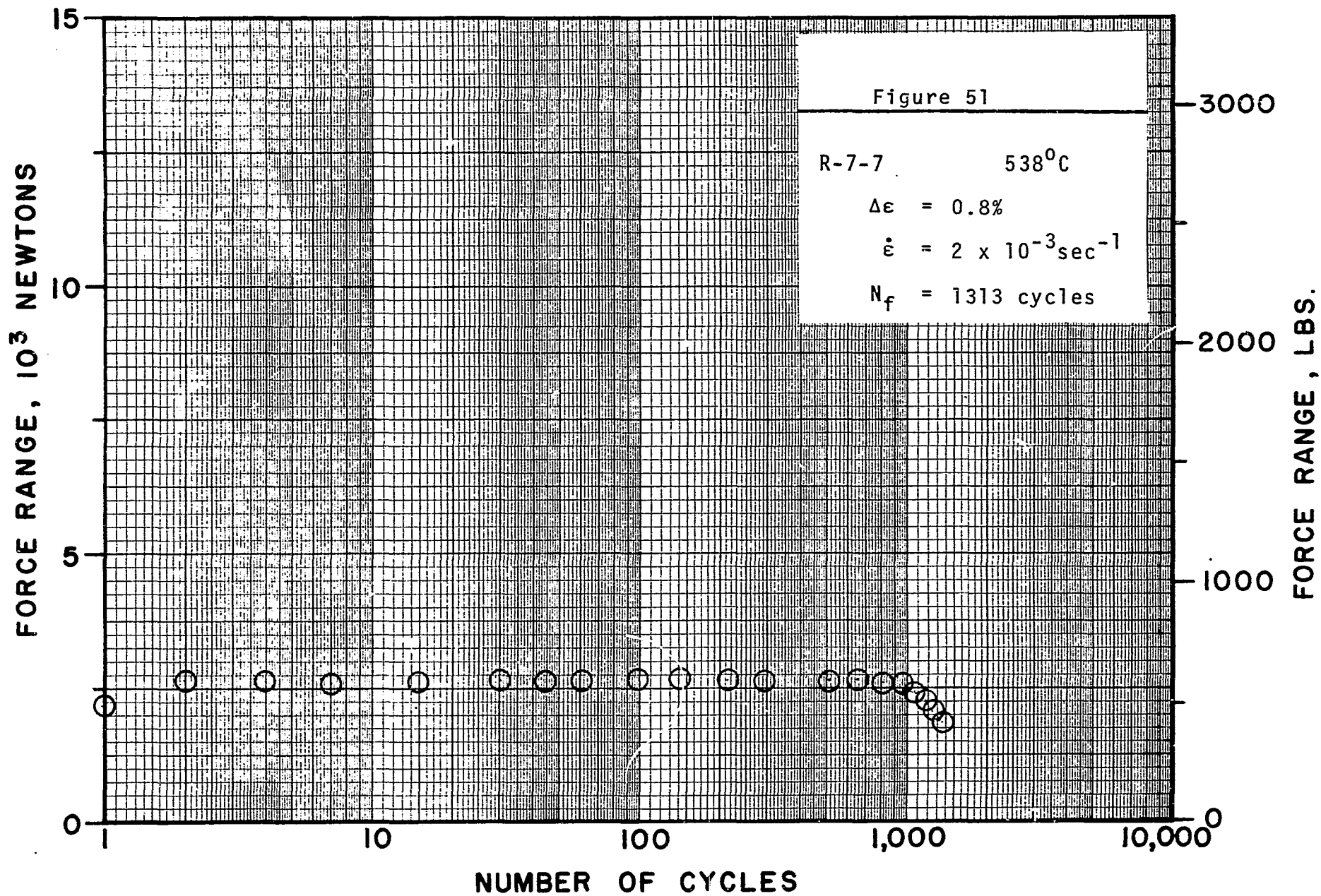












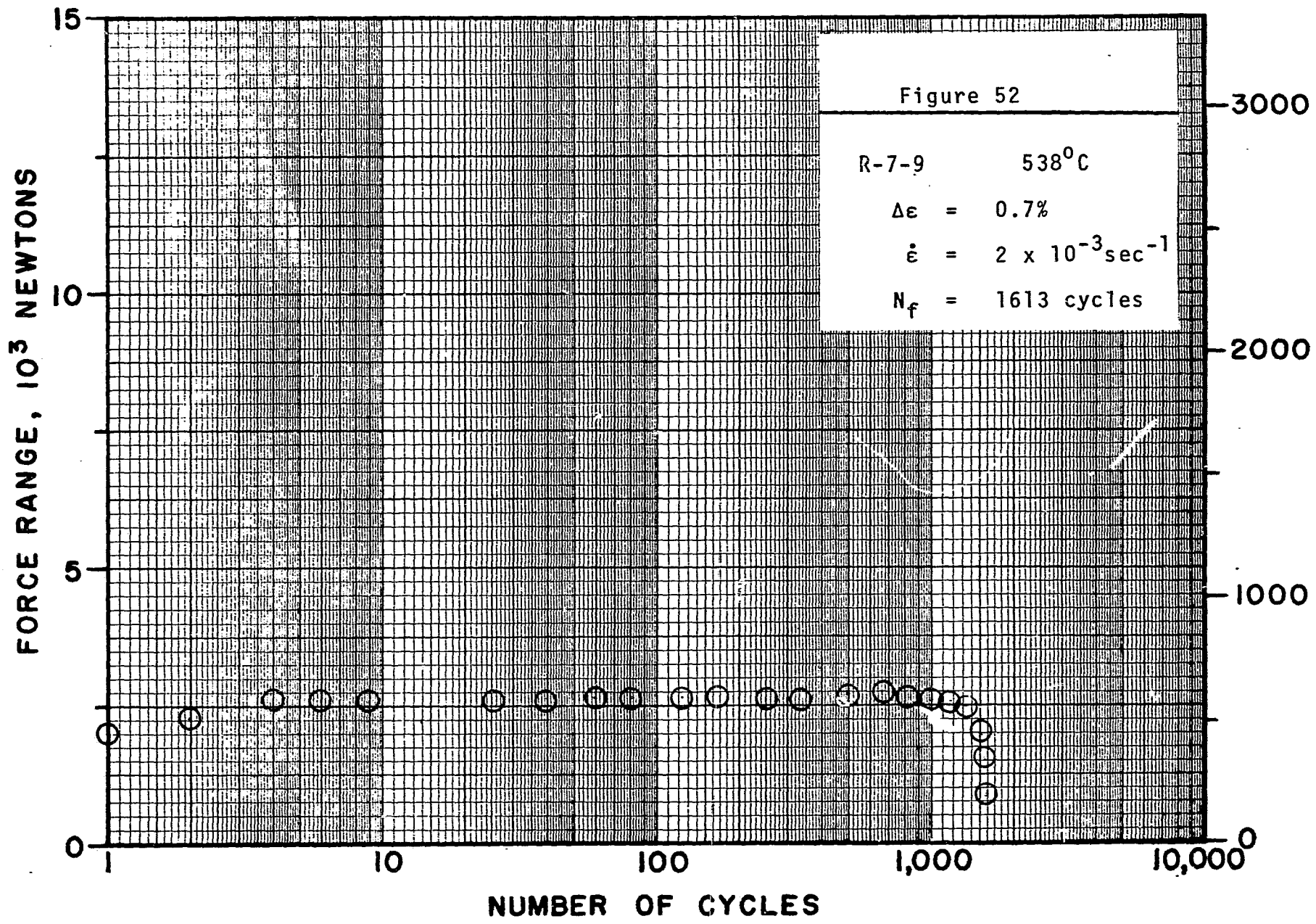


Figure 53

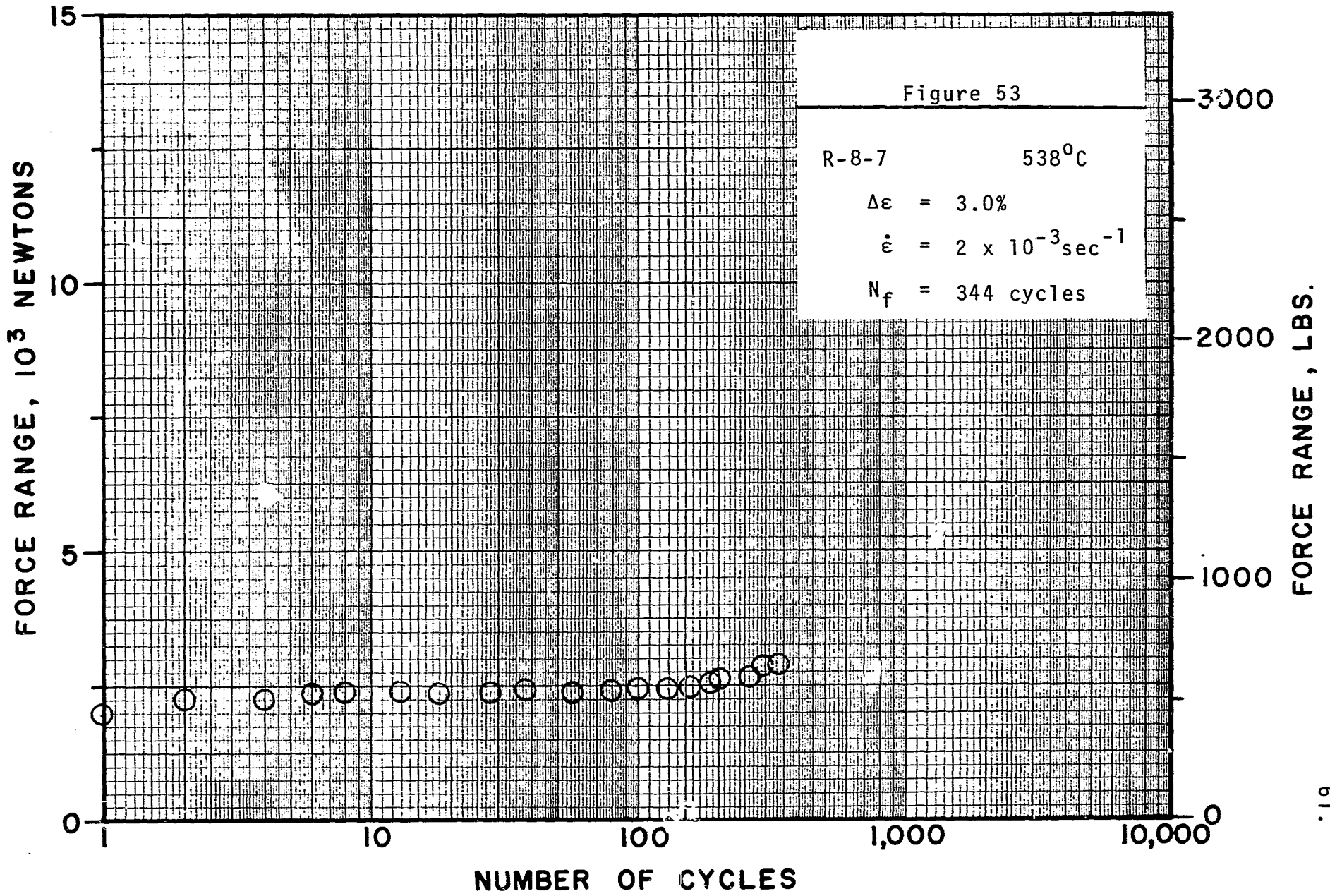
R-8-7

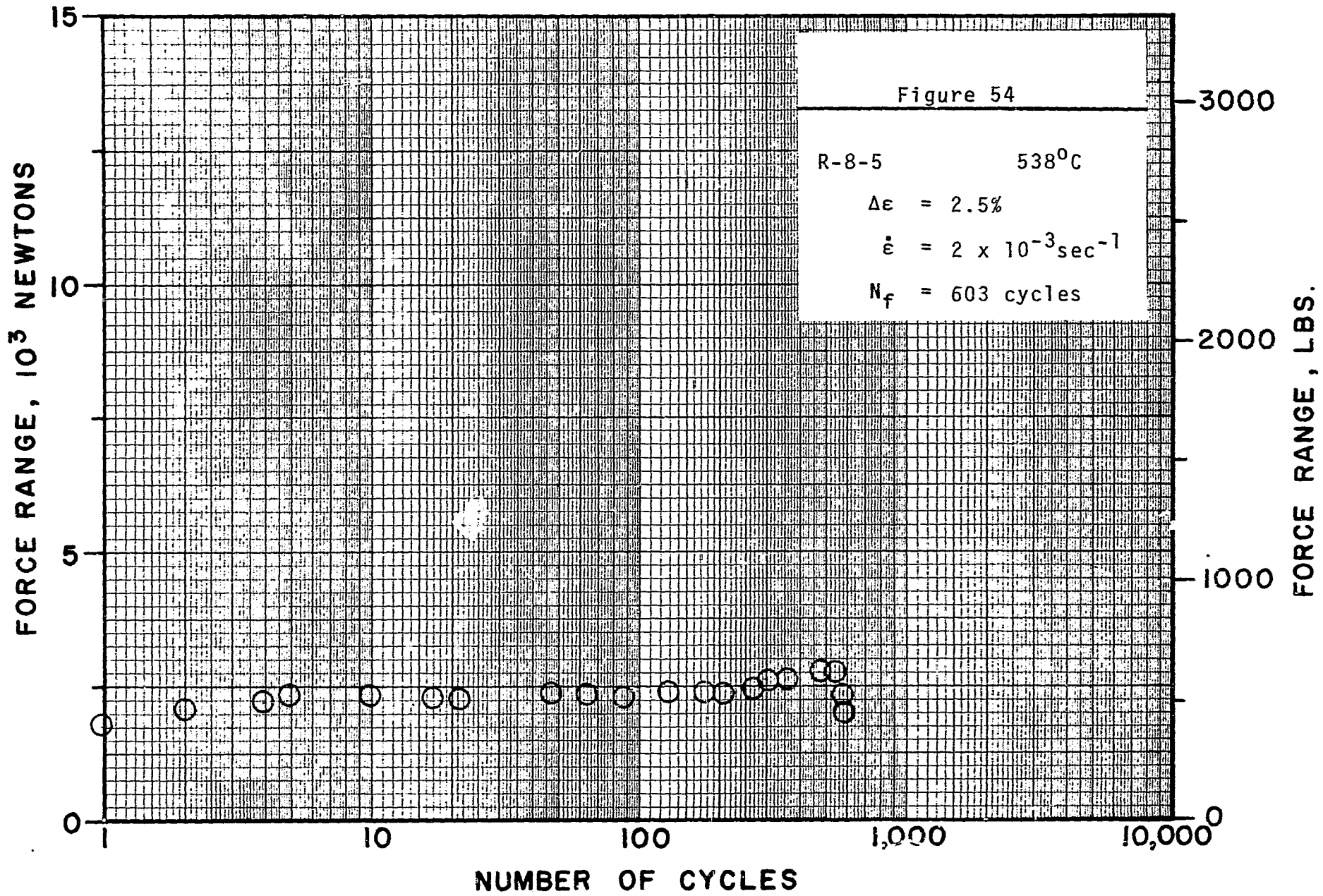
538⁰C

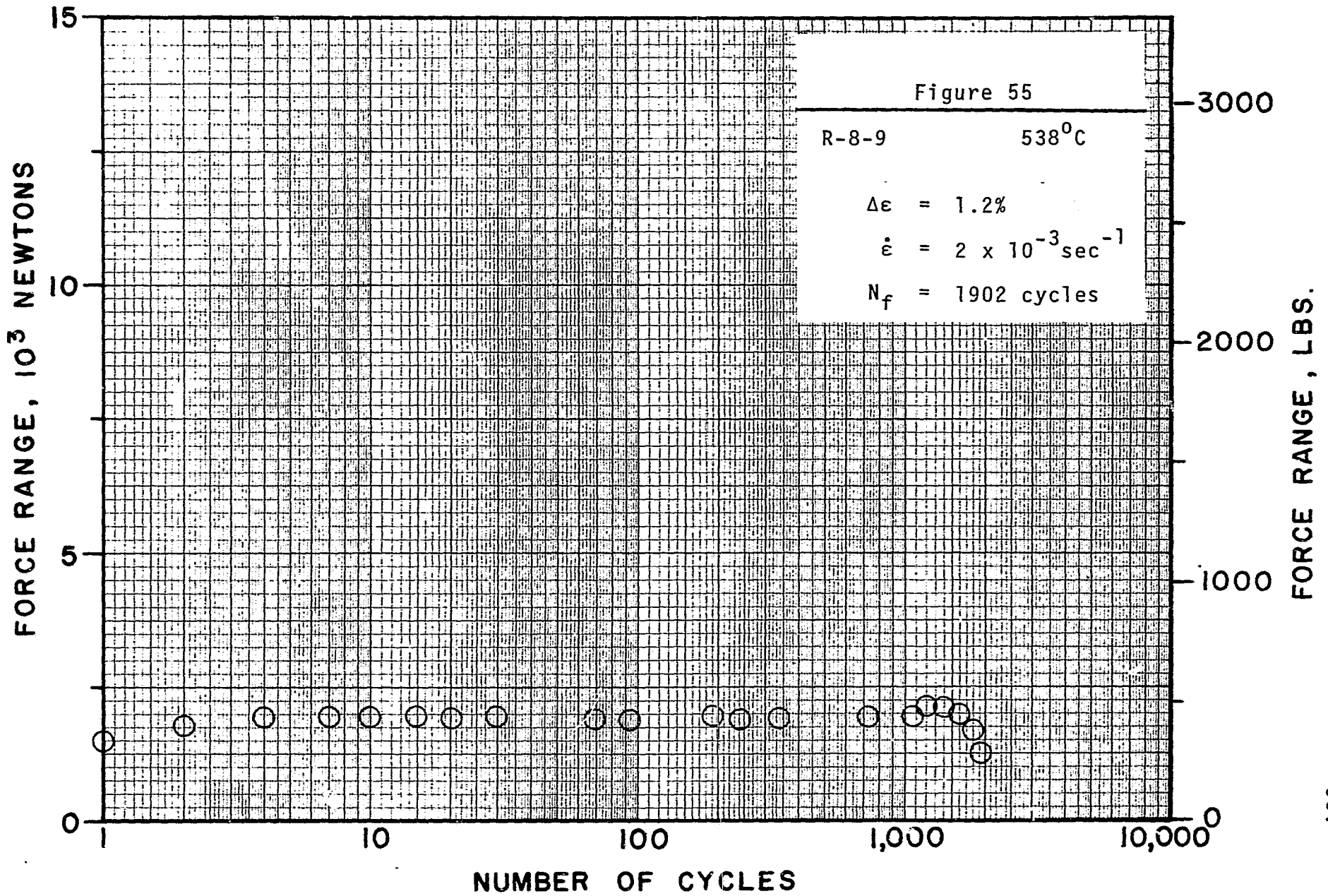
$\Delta\epsilon = 3.0\%$

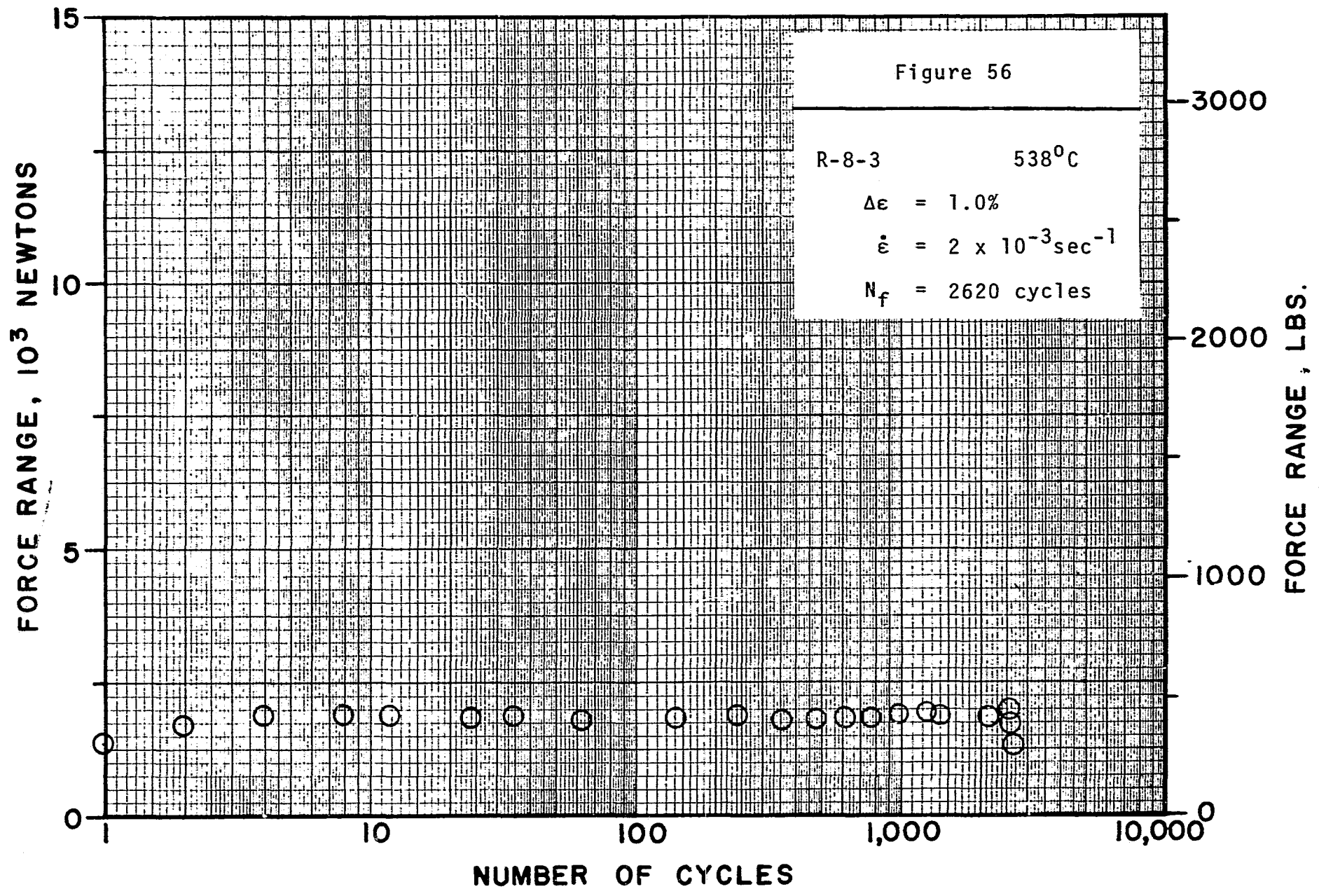
$\dot{\epsilon} = 2 \times 10^{-3} \text{sec}^{-1}$

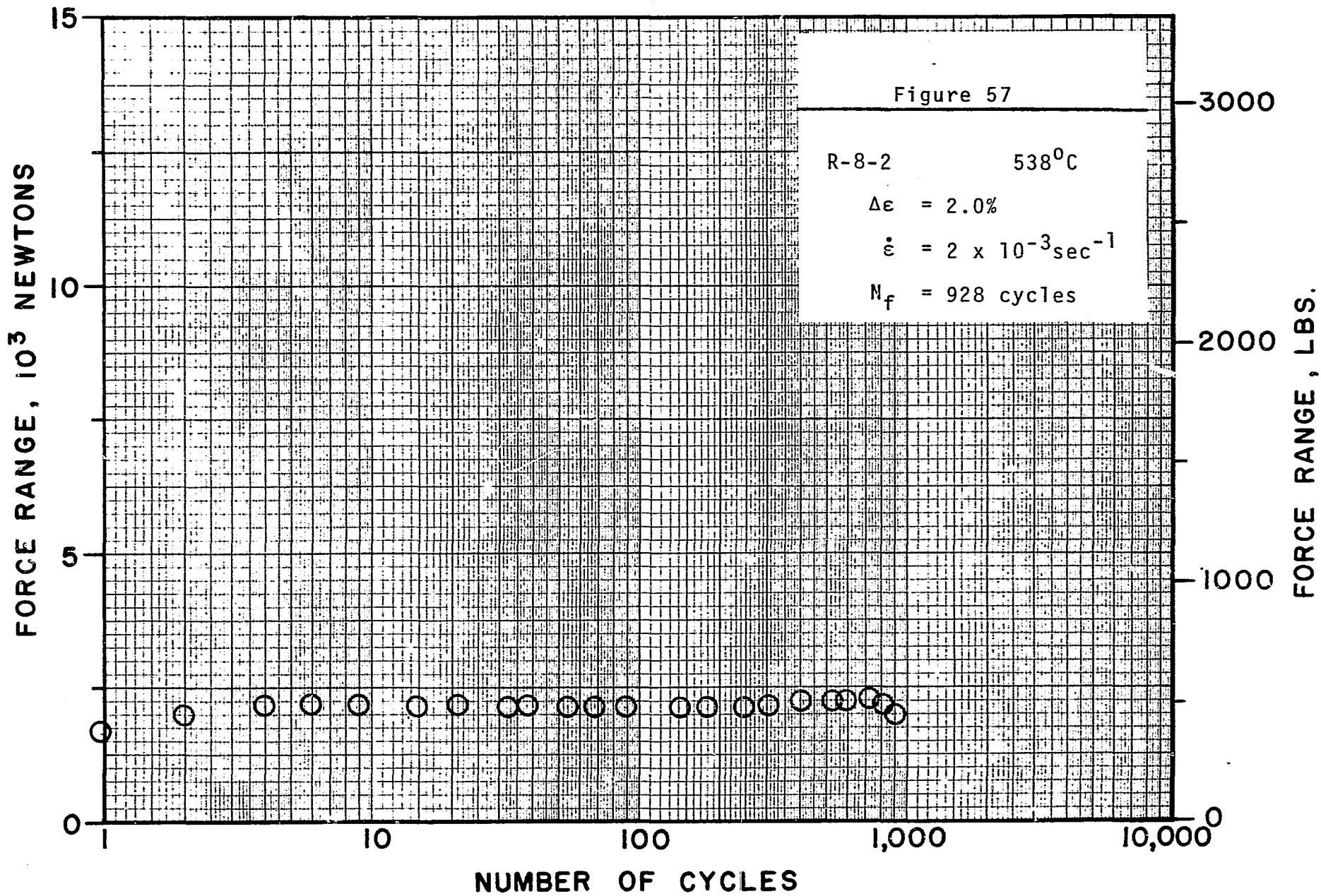
$N_f = 344 \text{ cycles}$

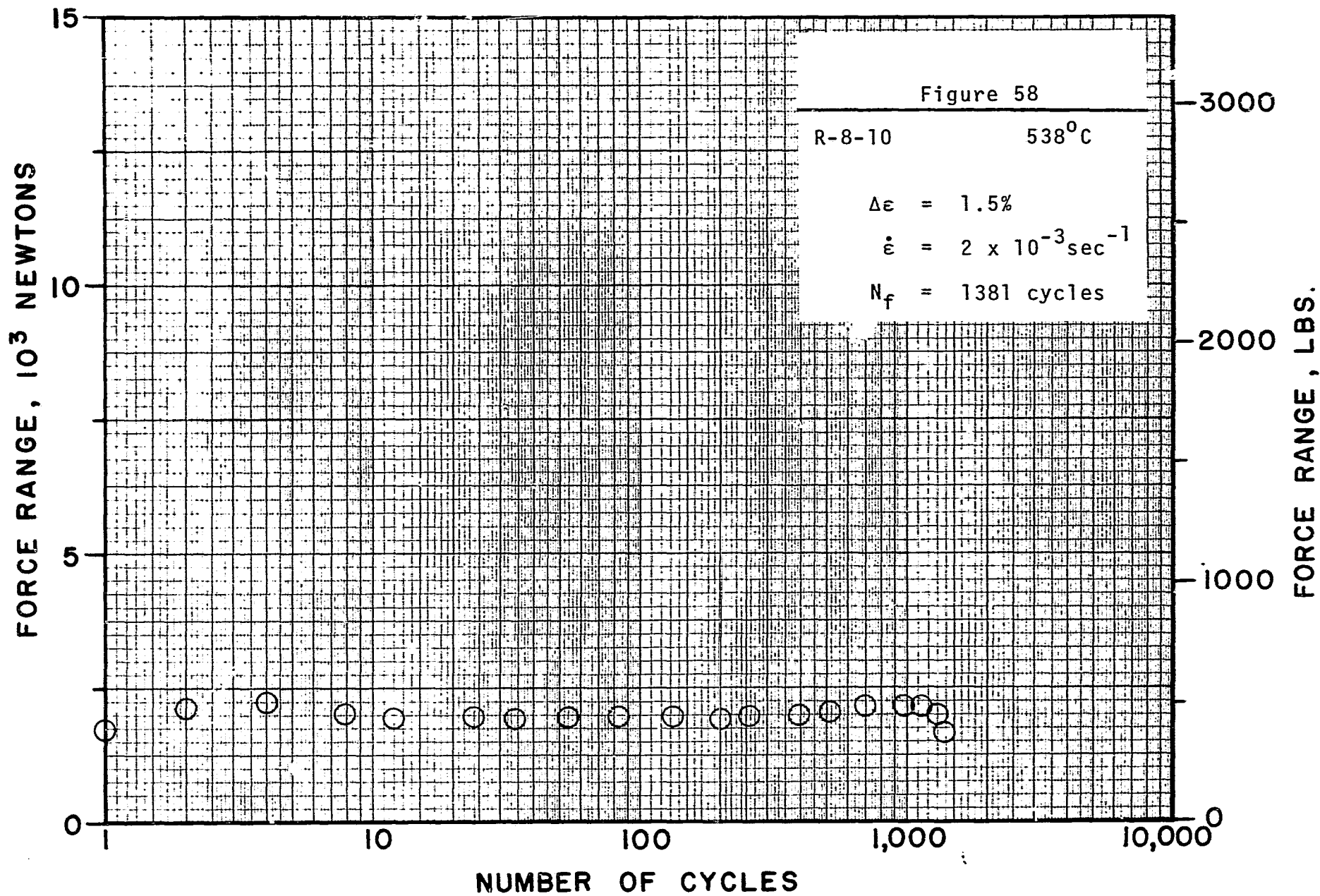


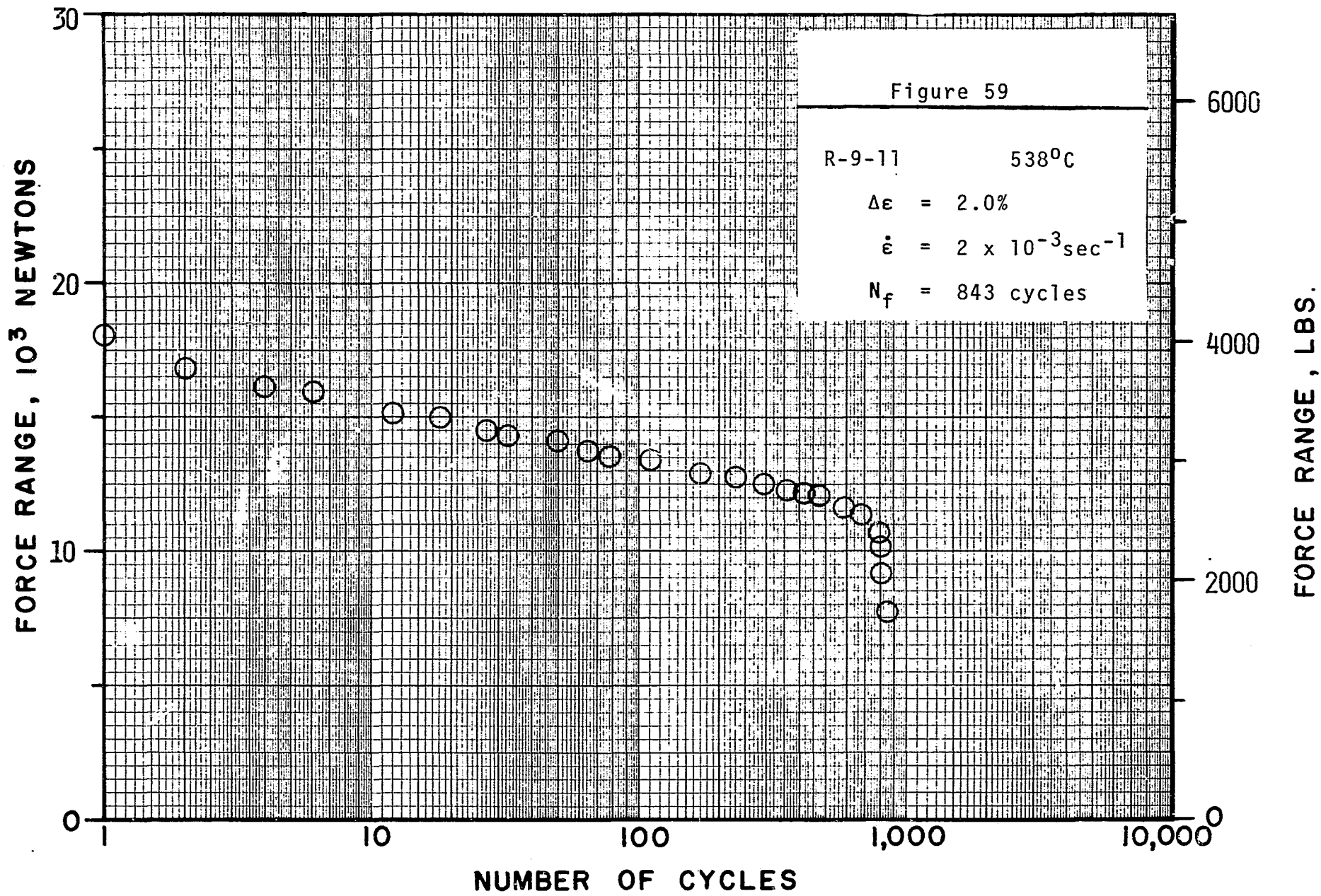


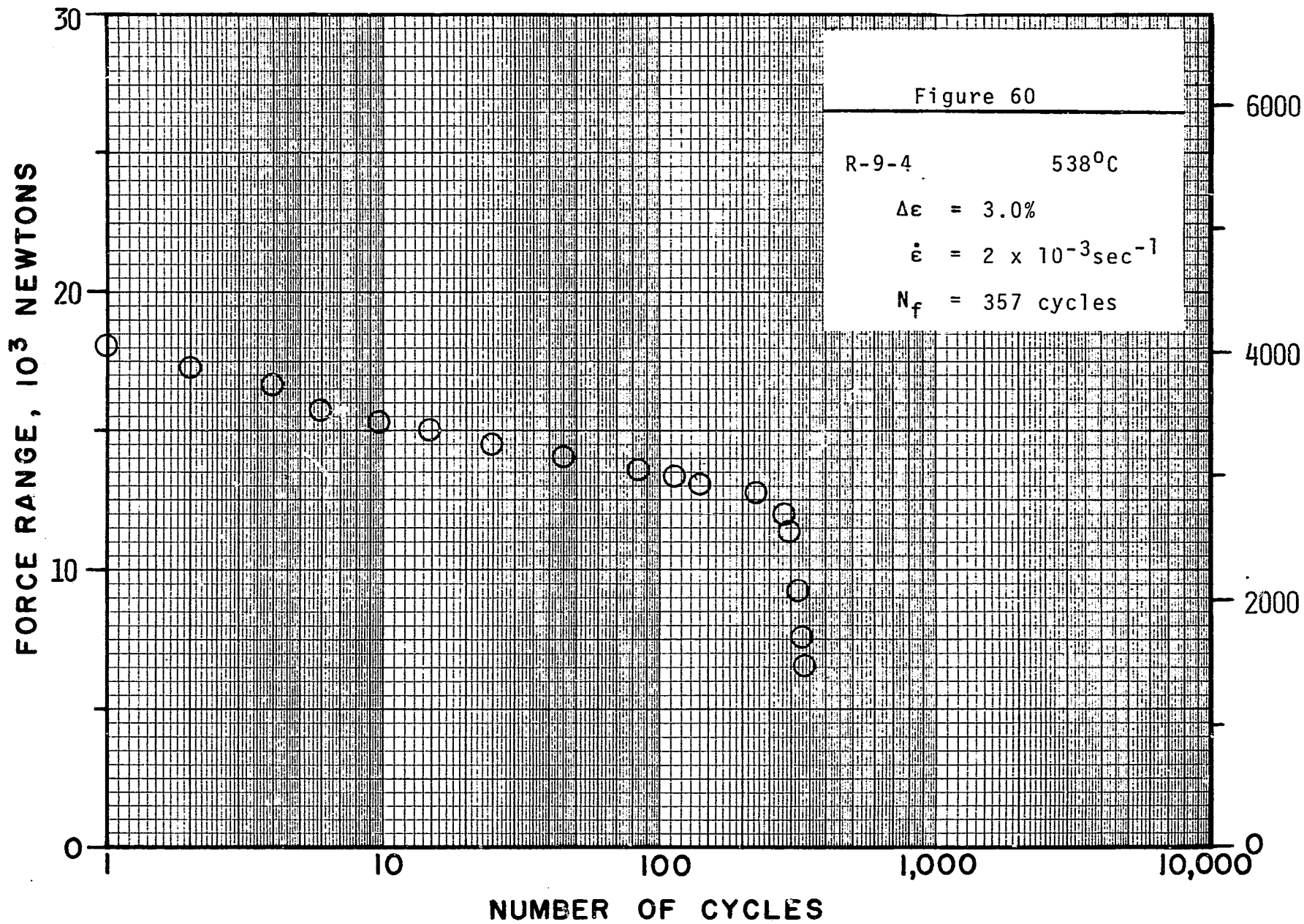


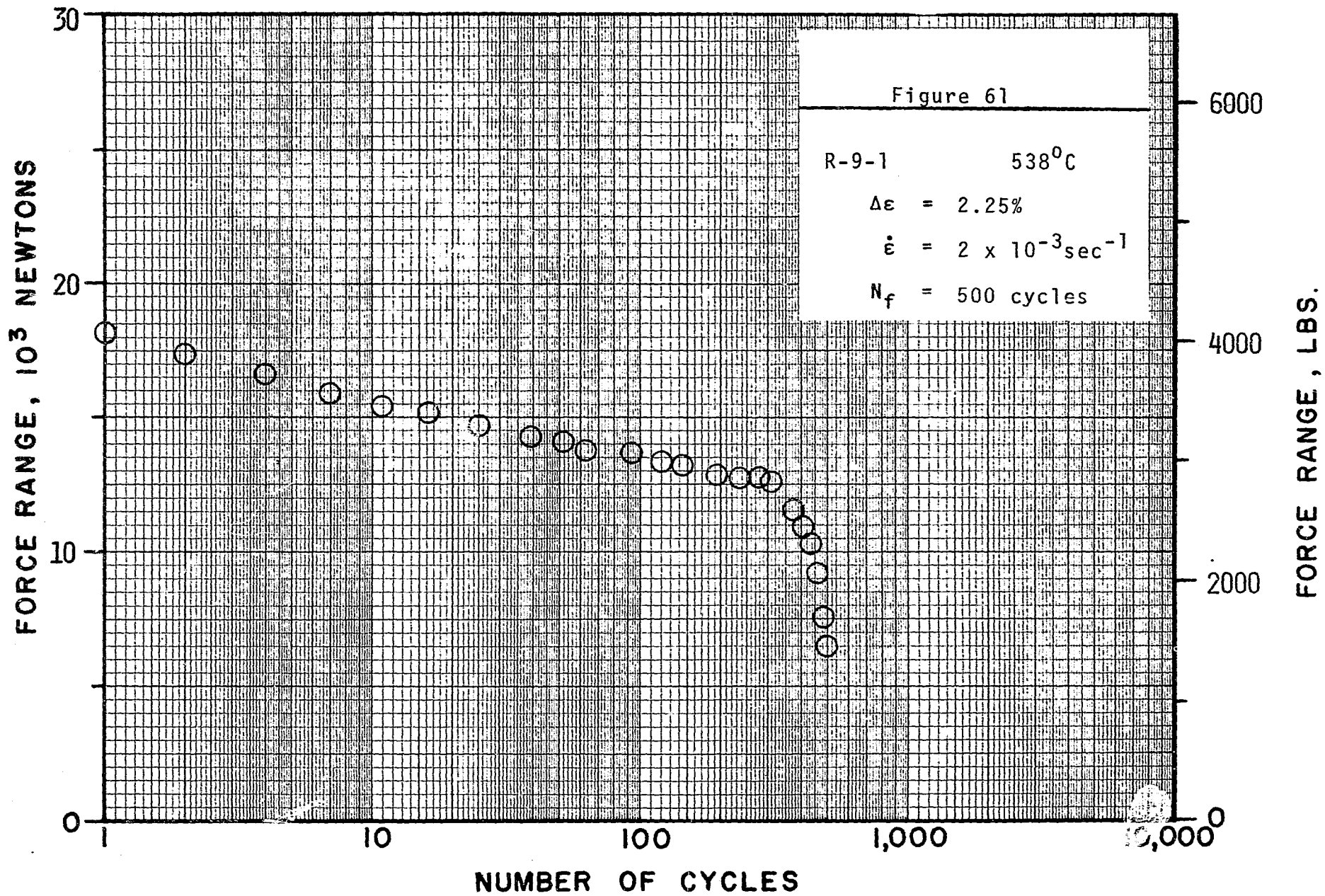


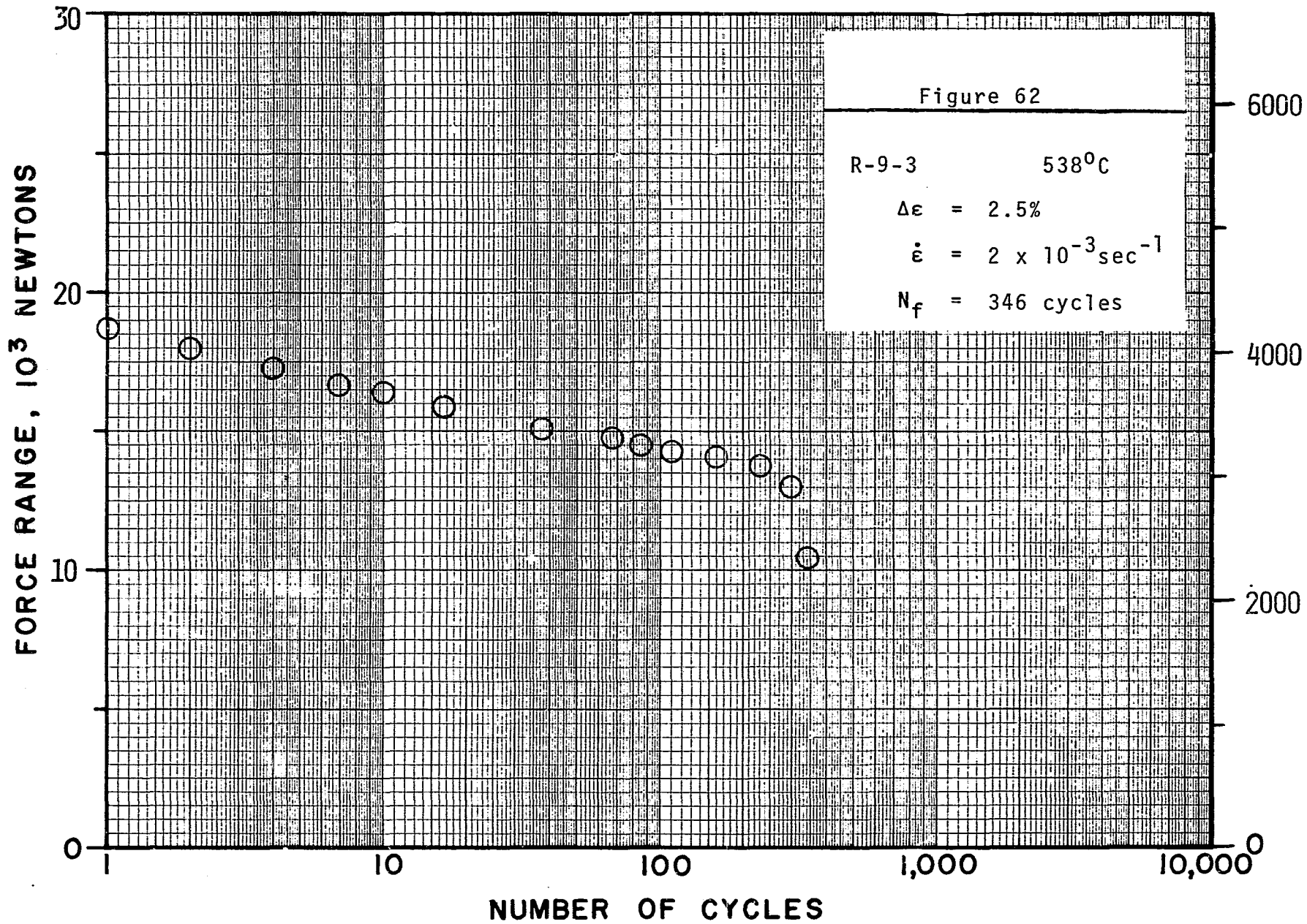




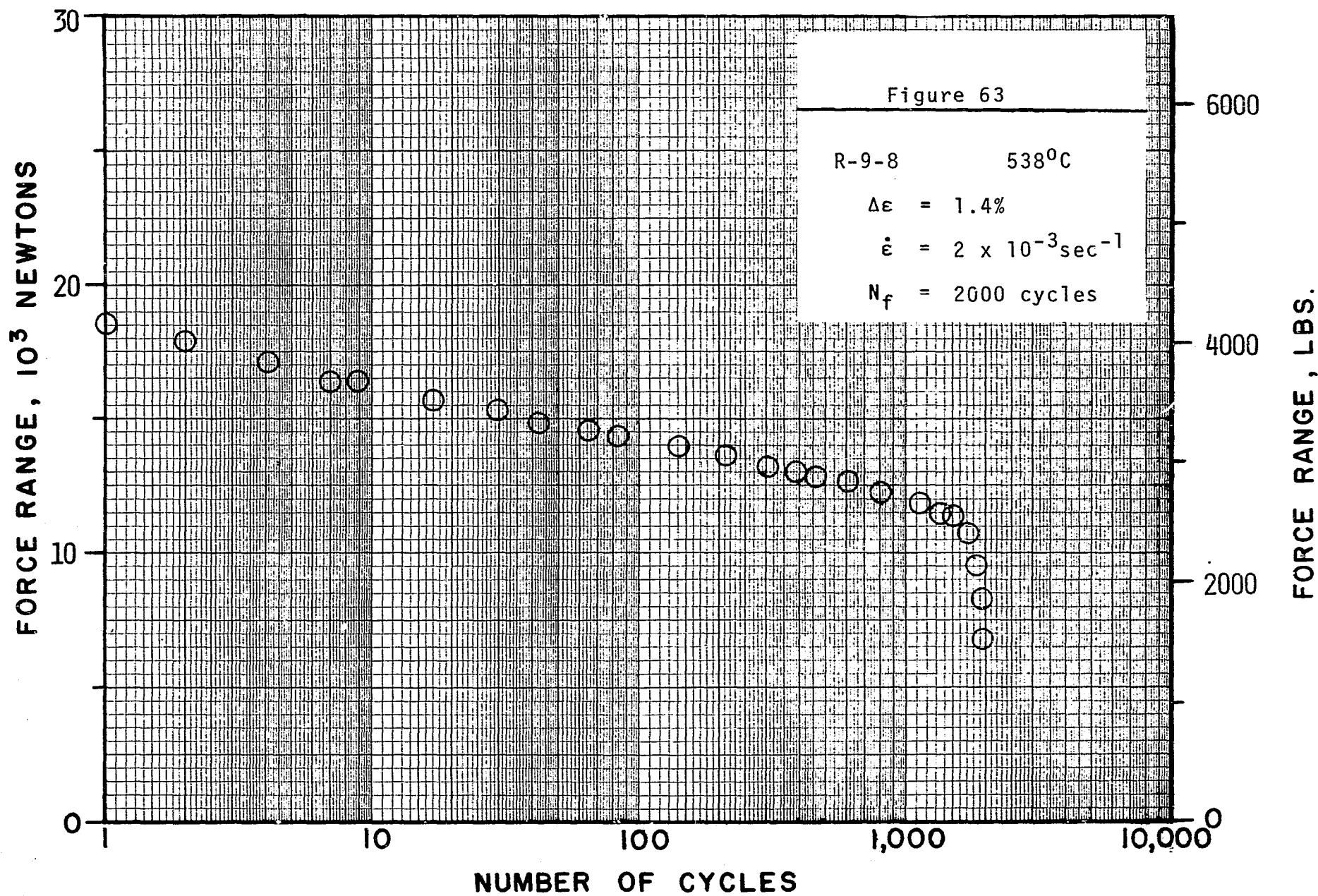


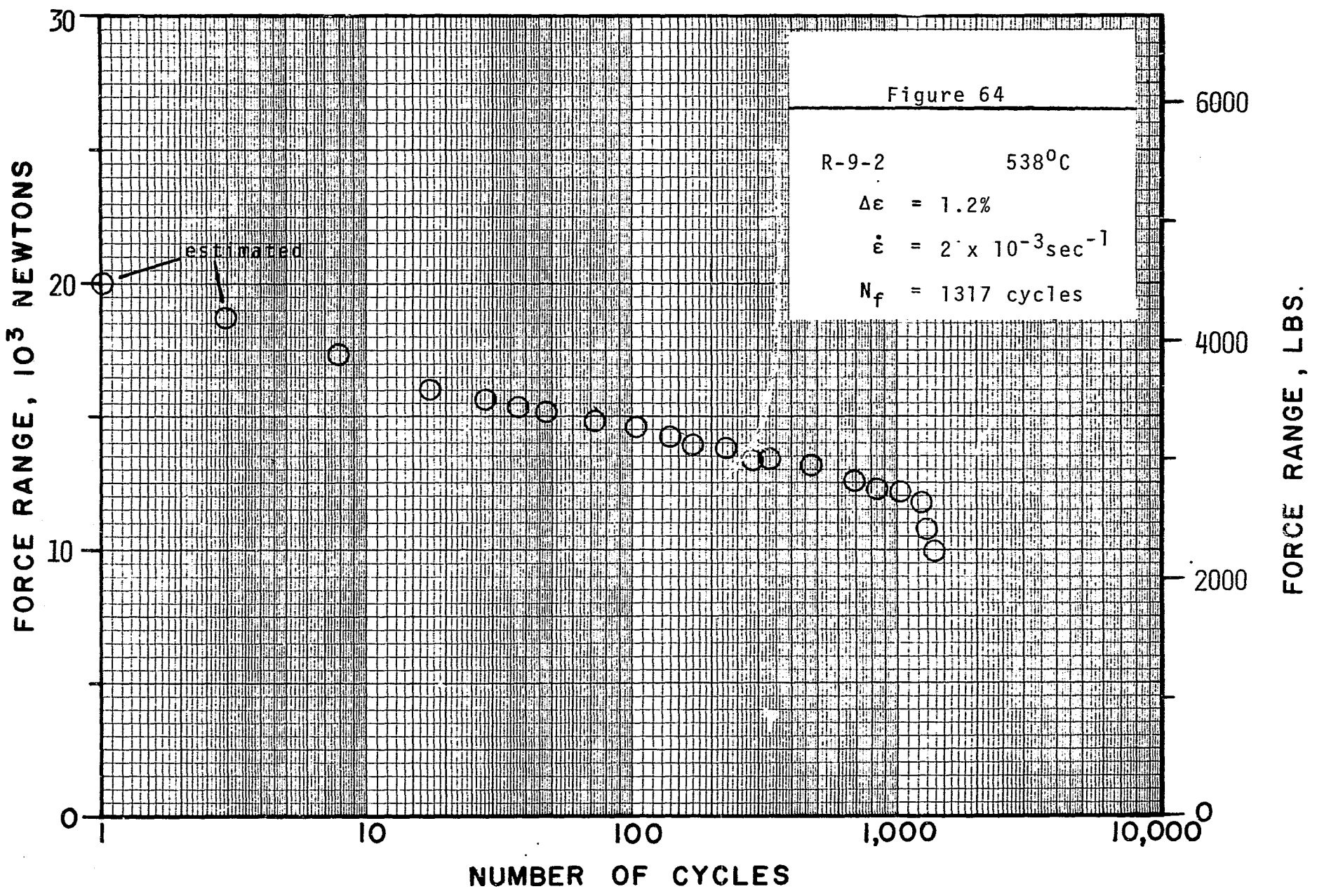


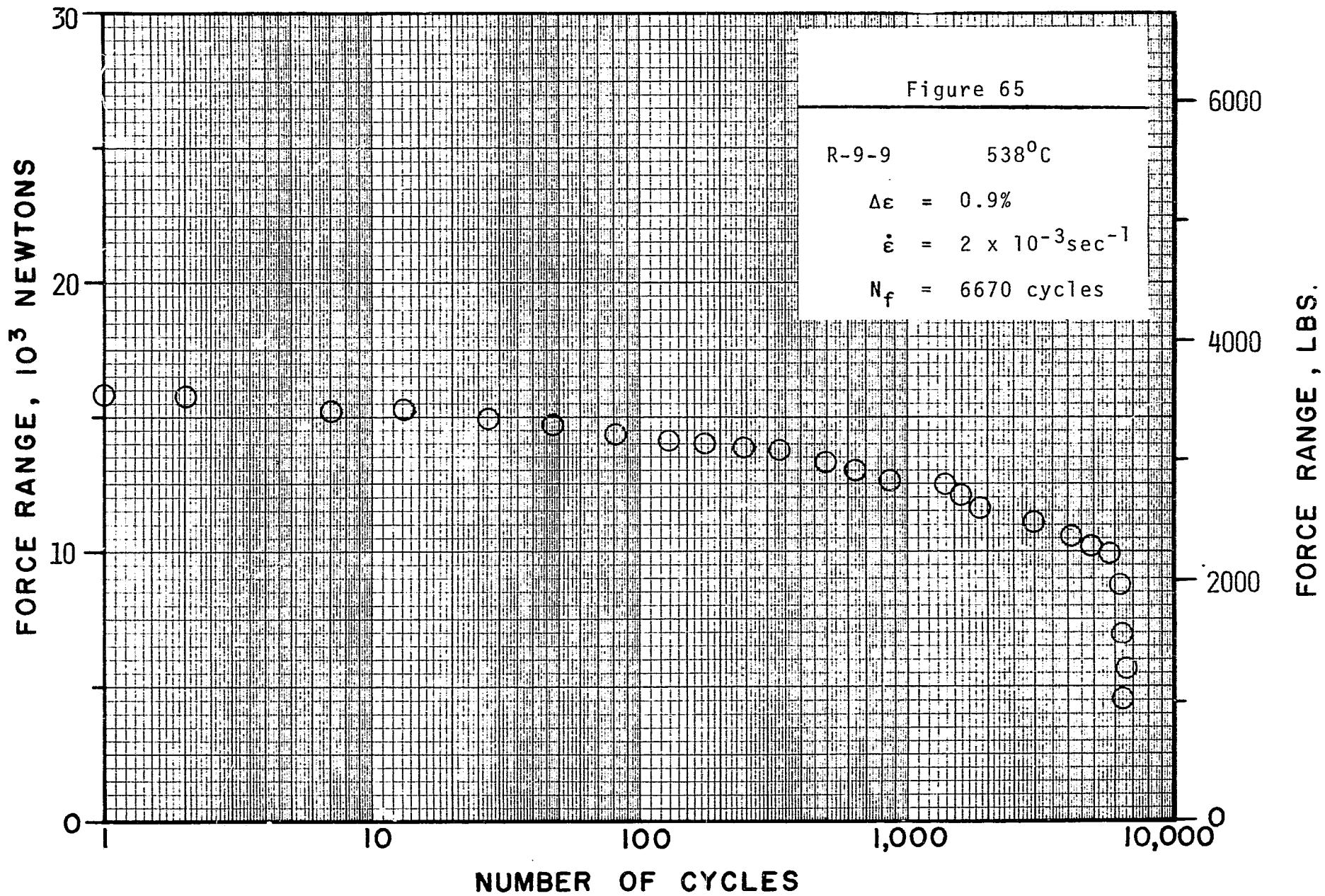


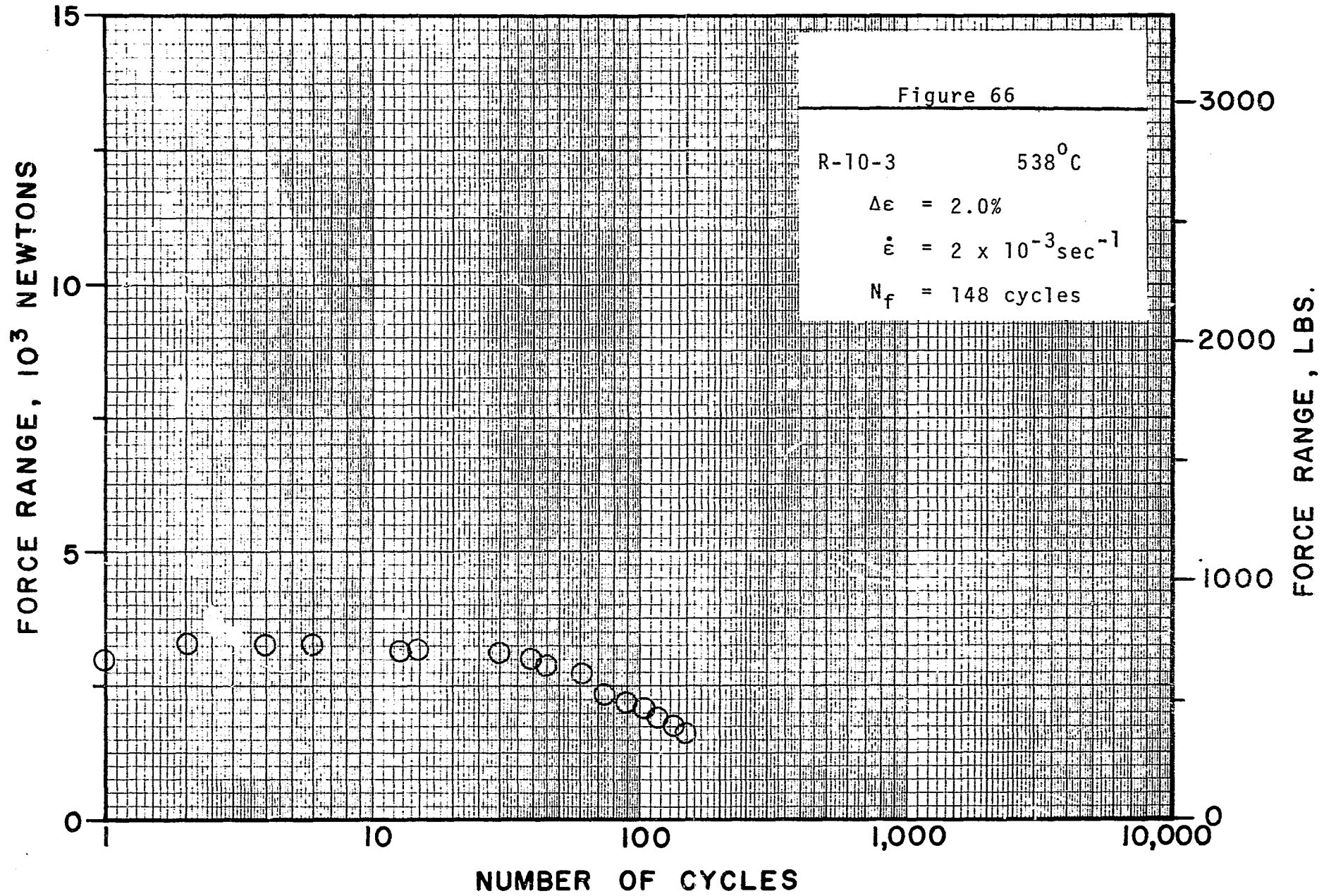


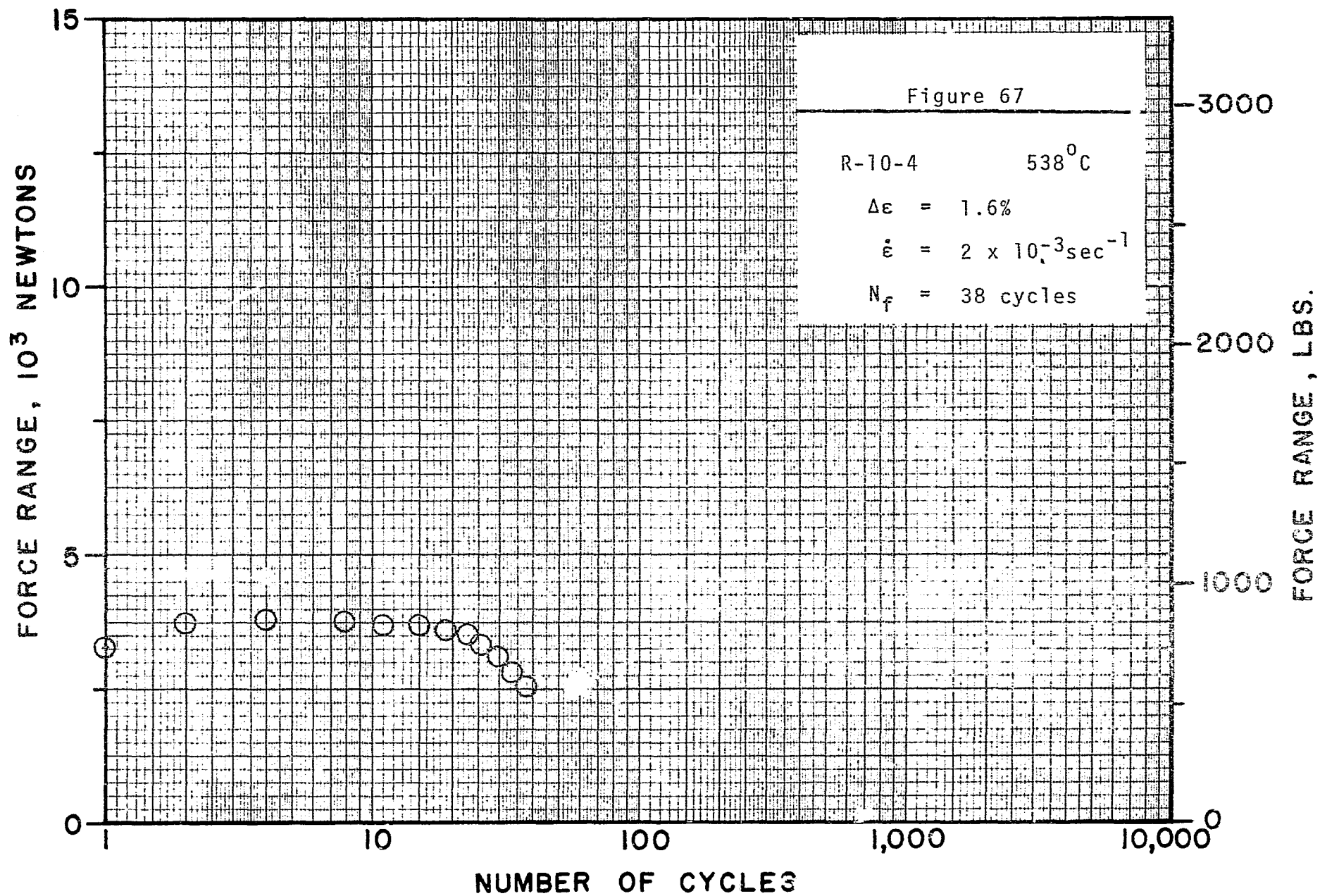
FORCE RANGE, LBS.

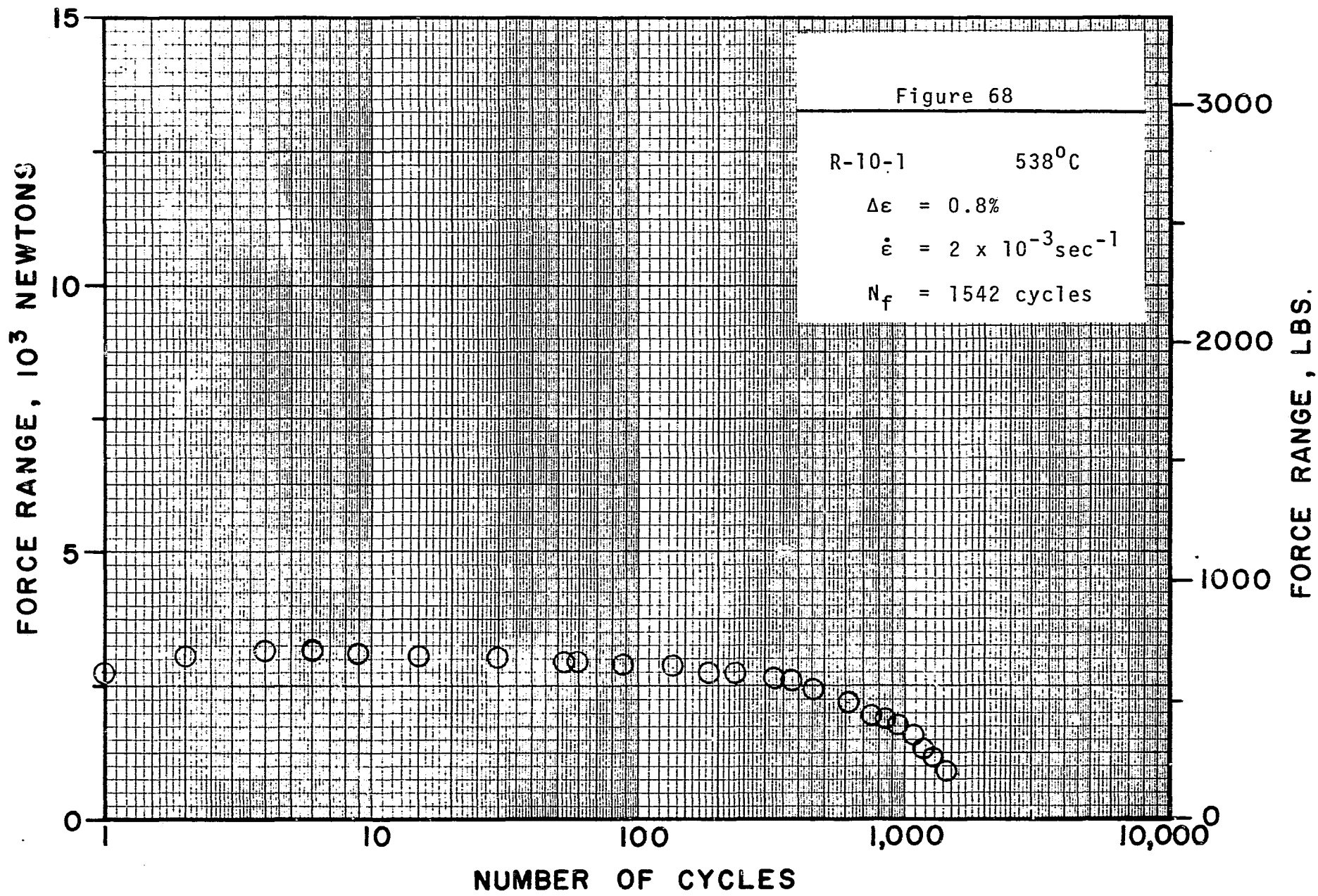


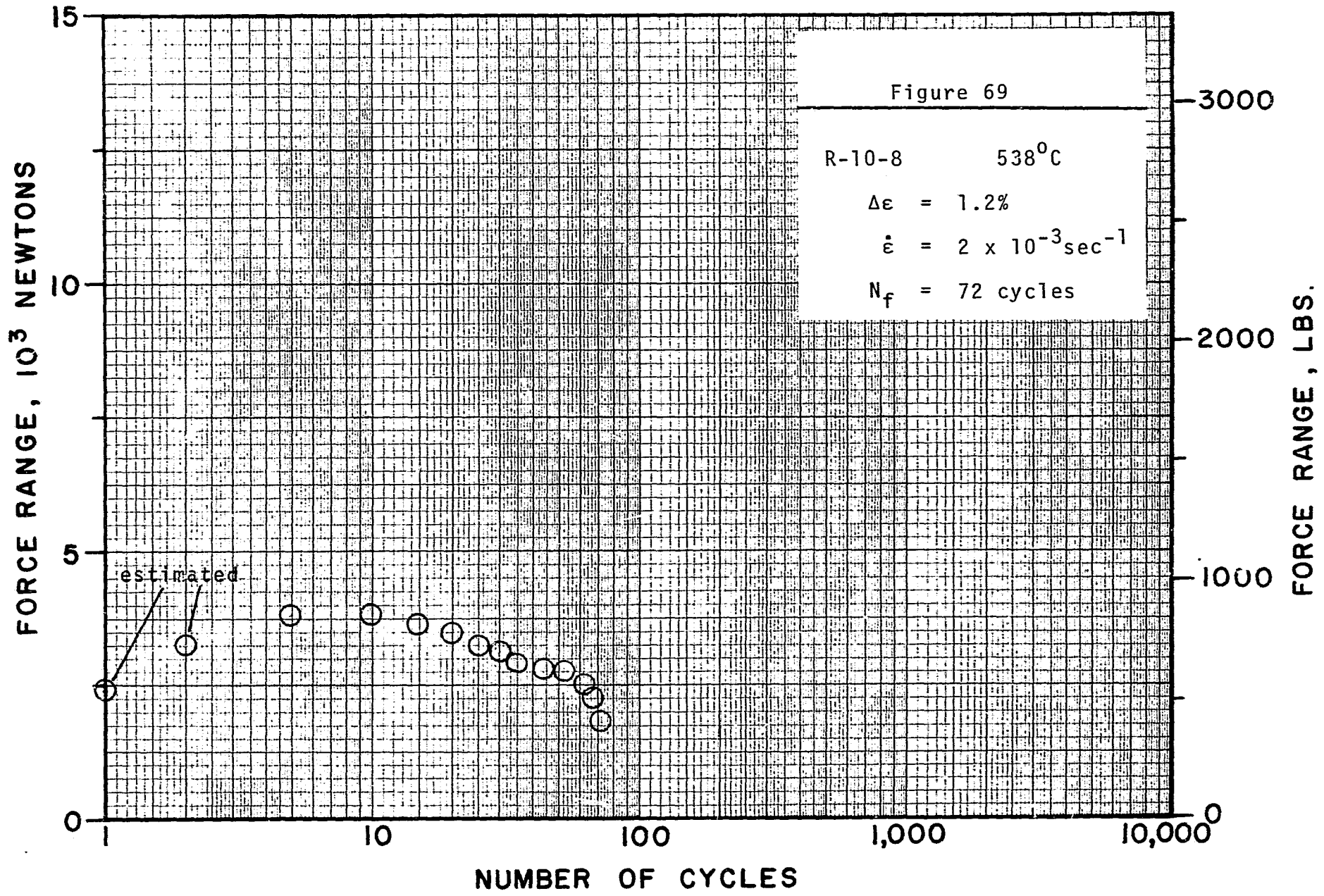


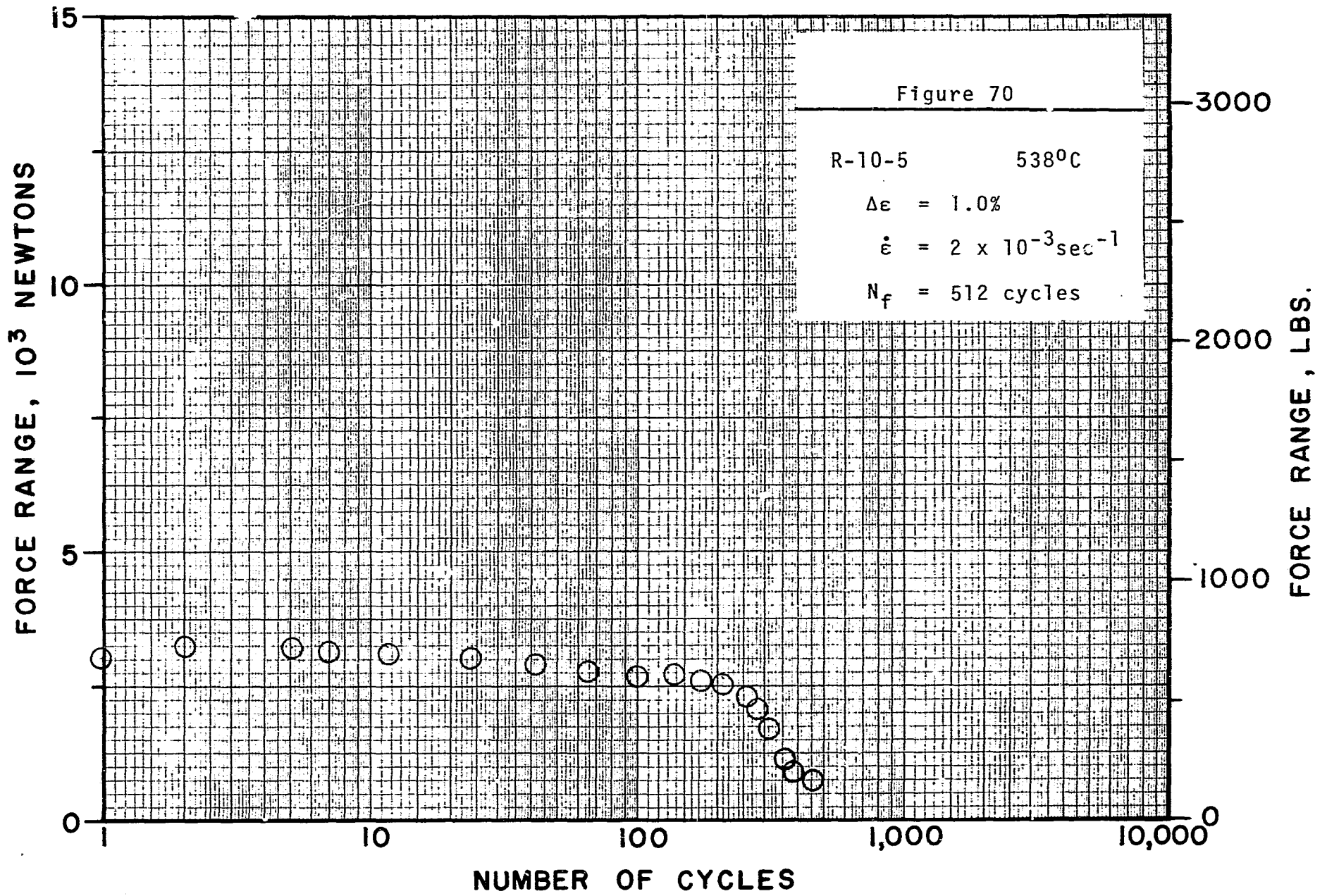


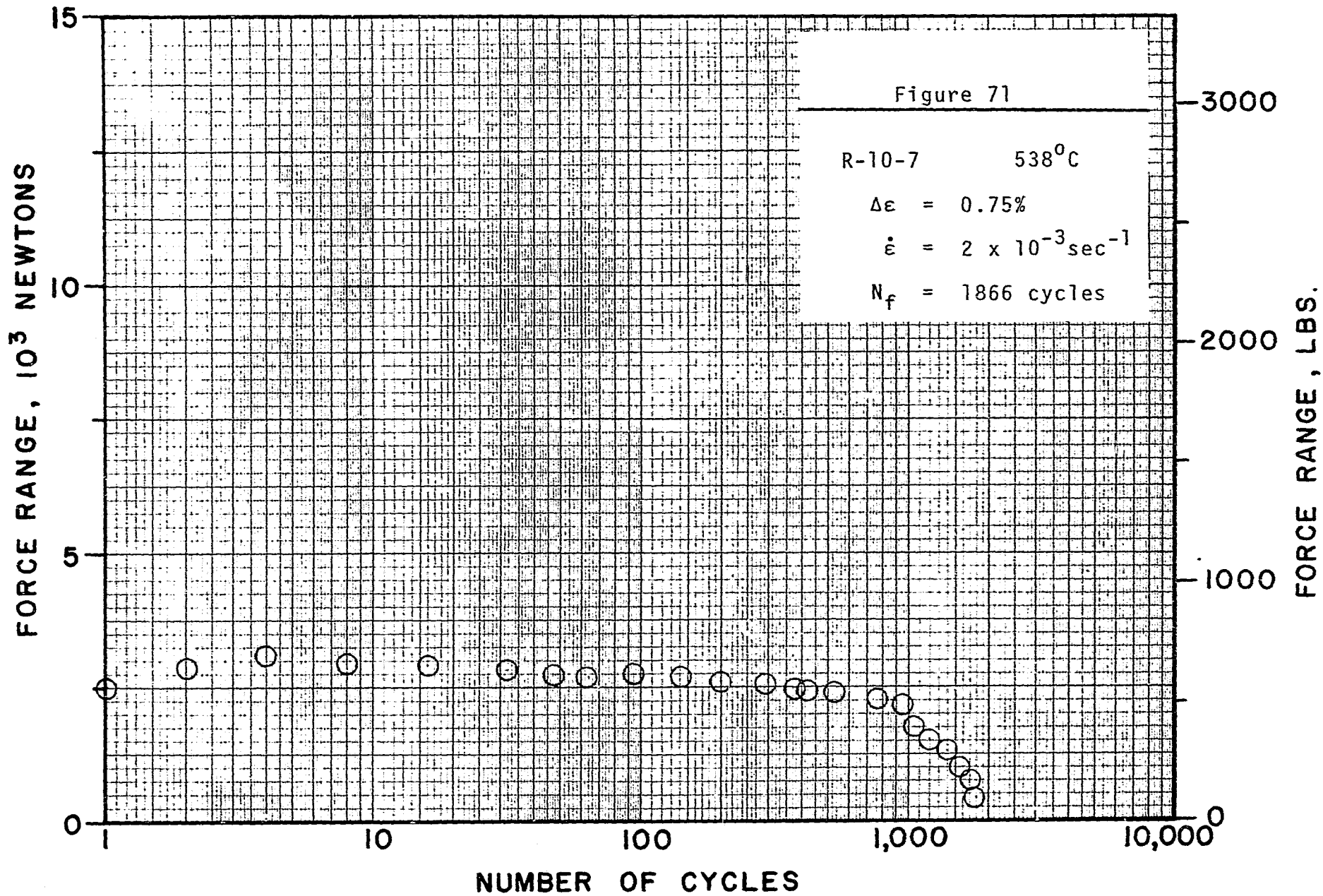


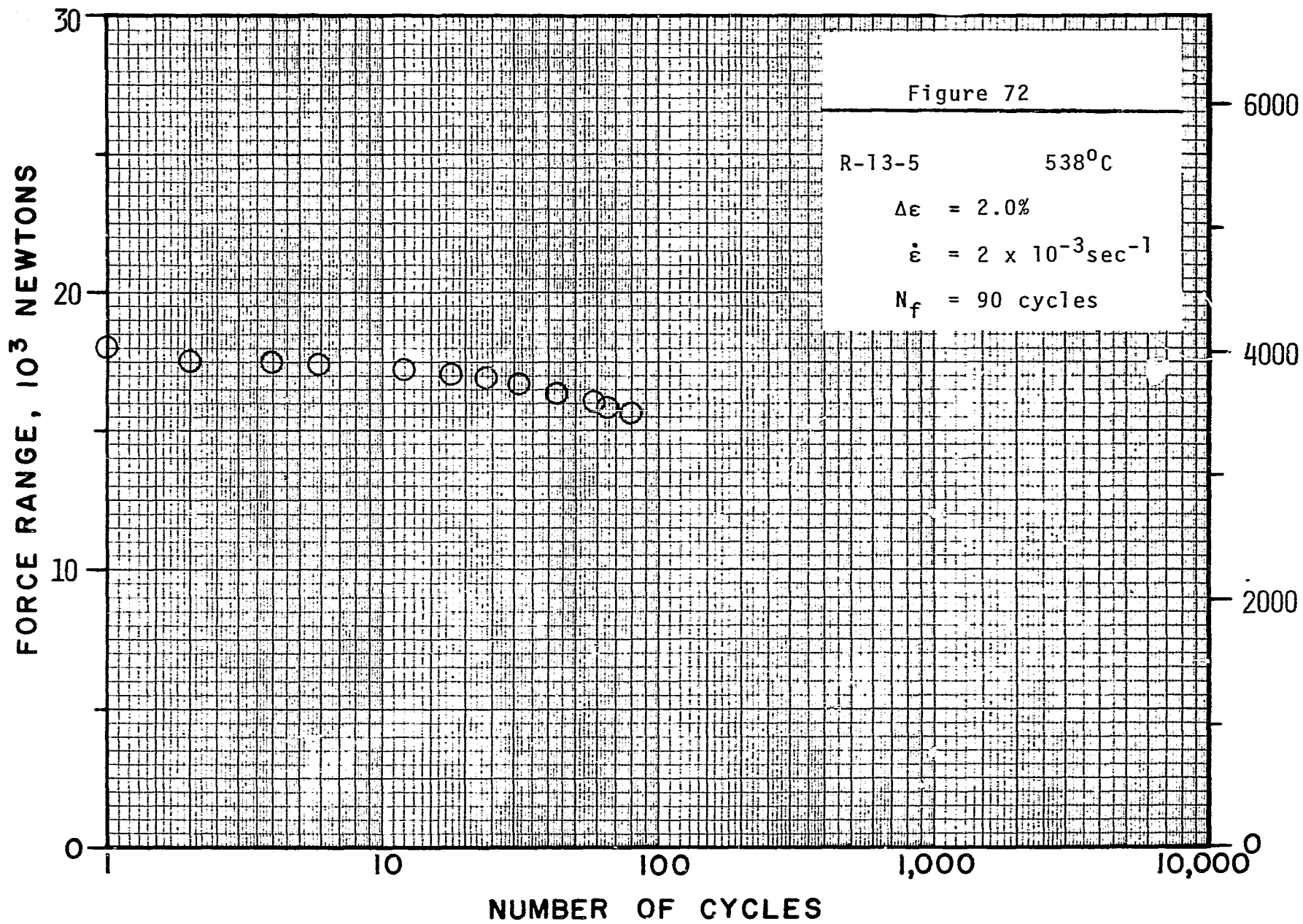




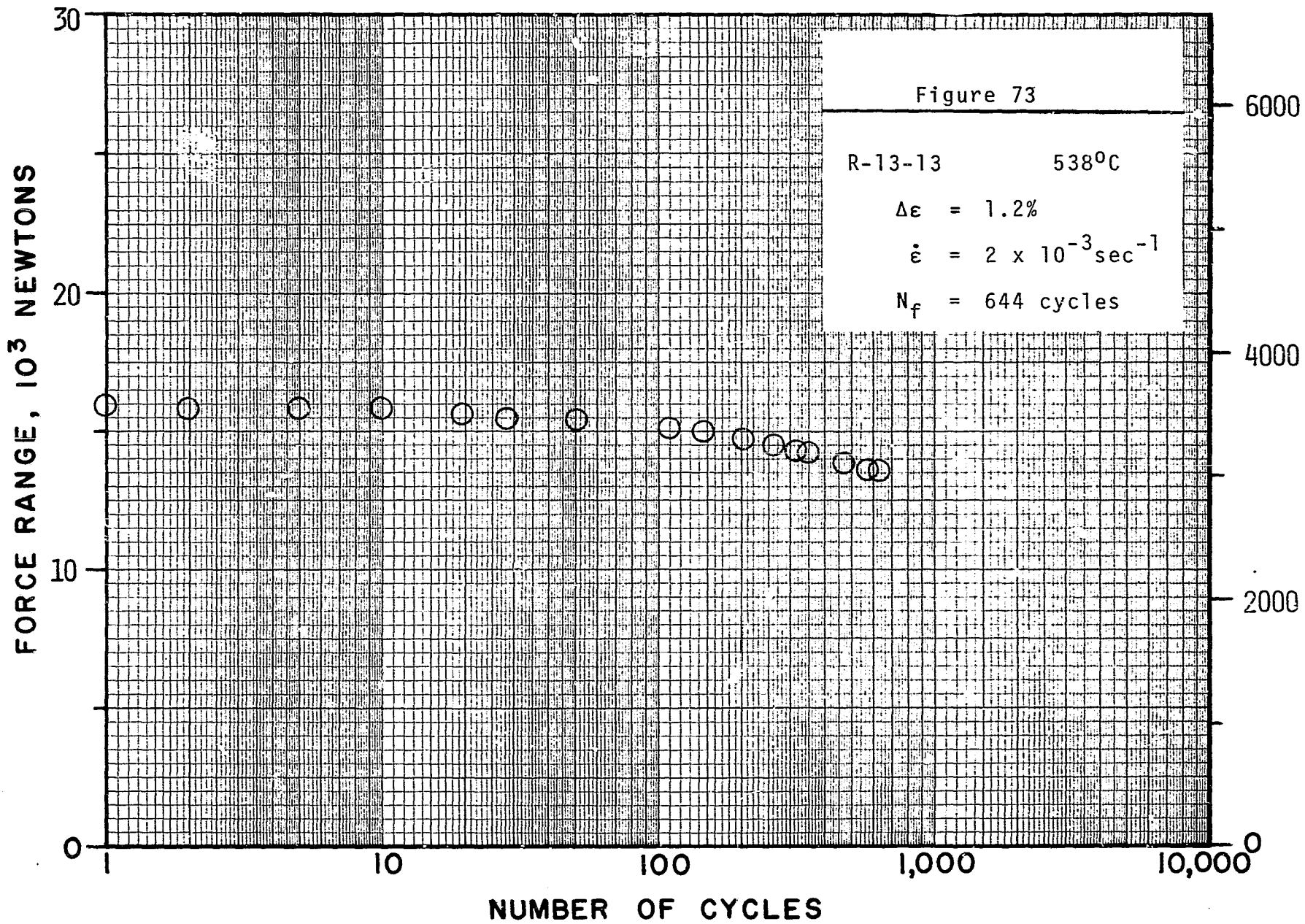


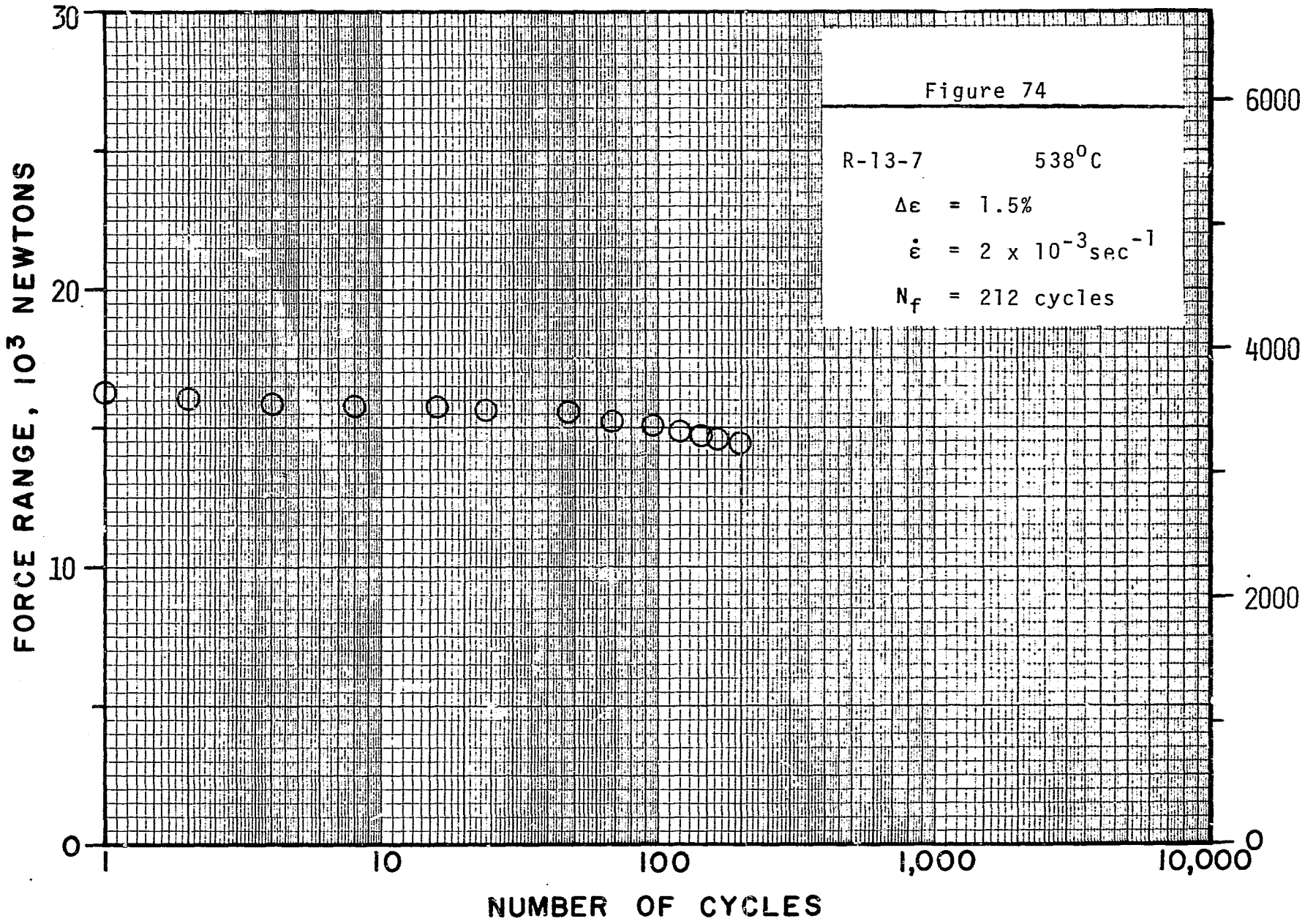


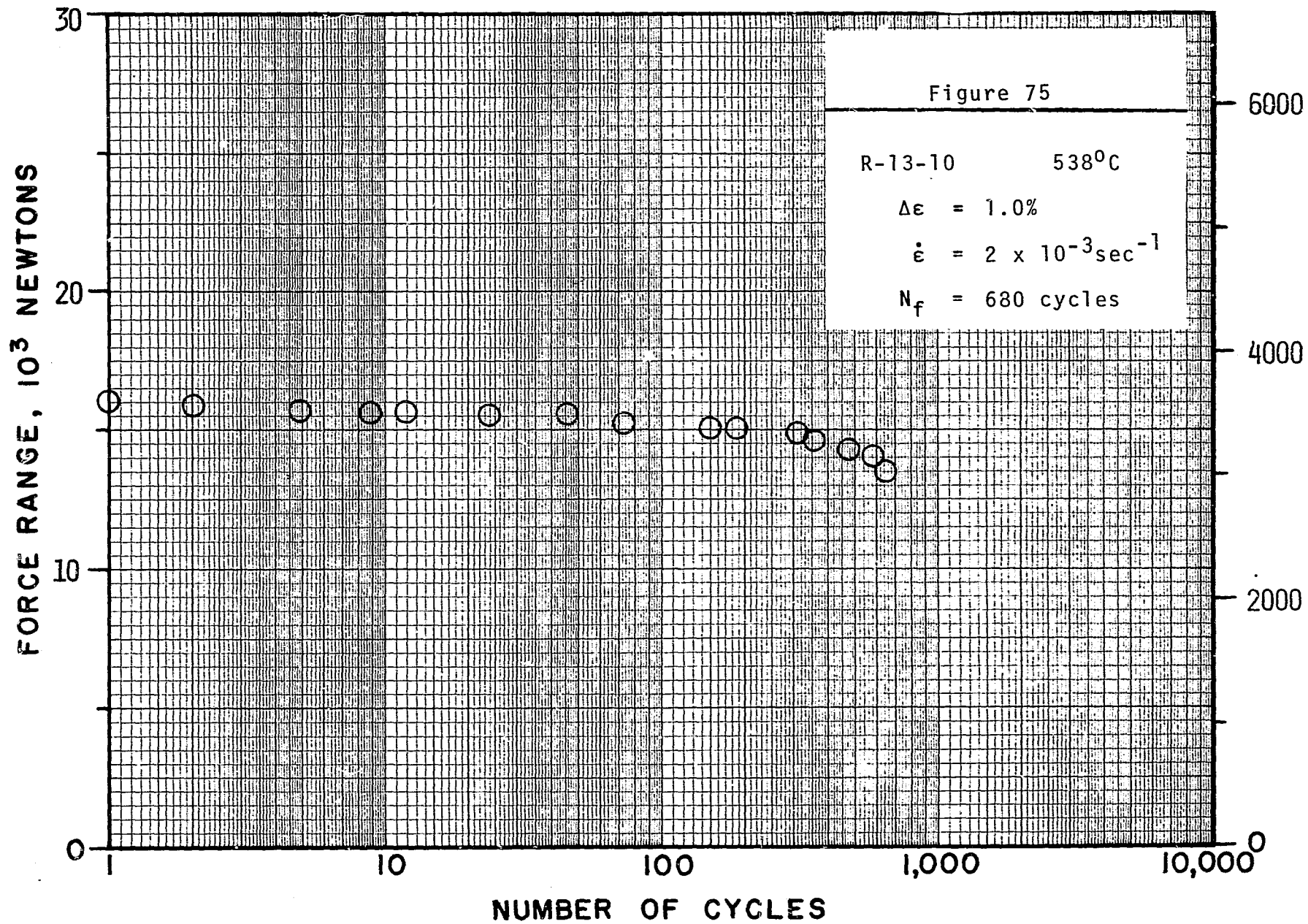


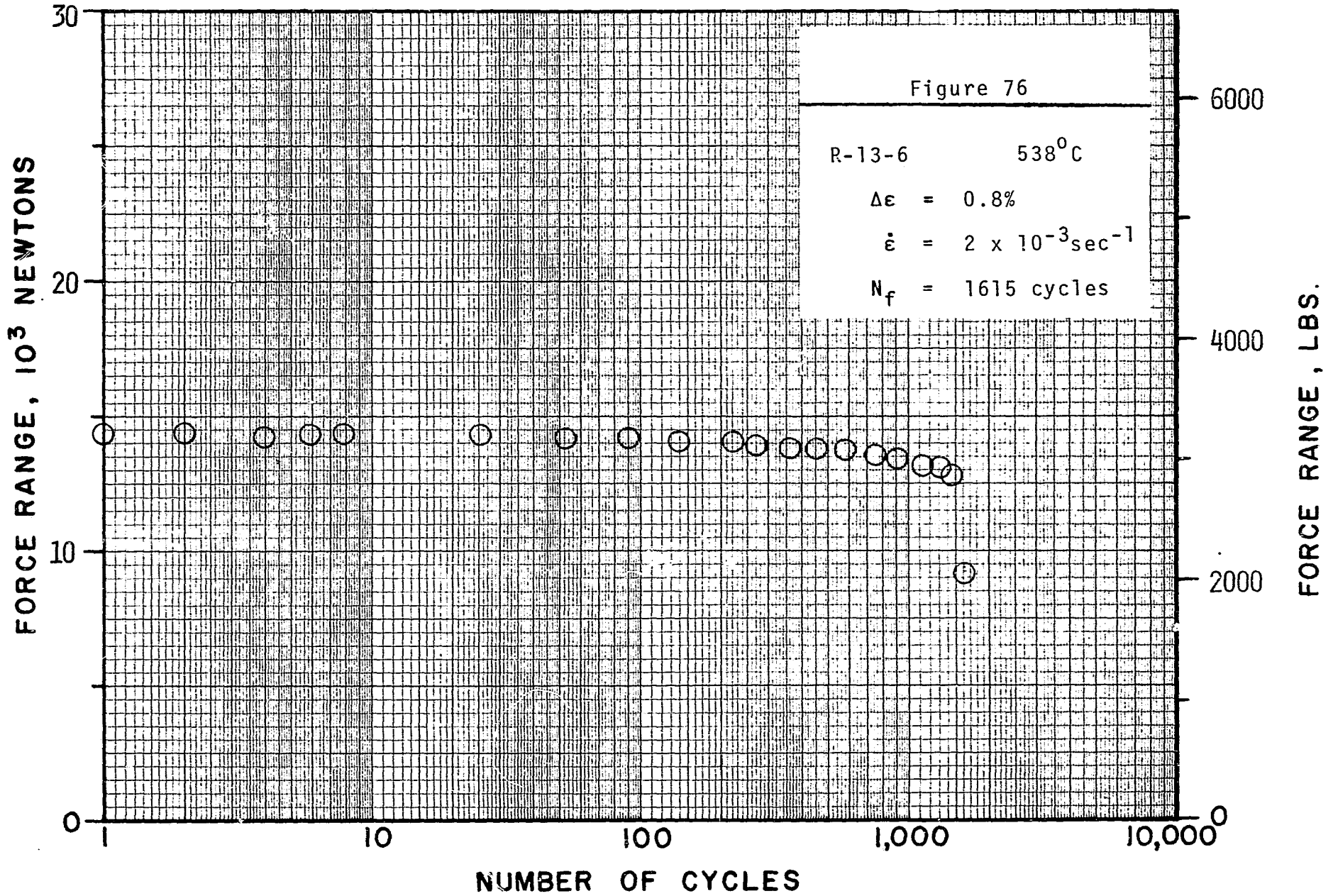


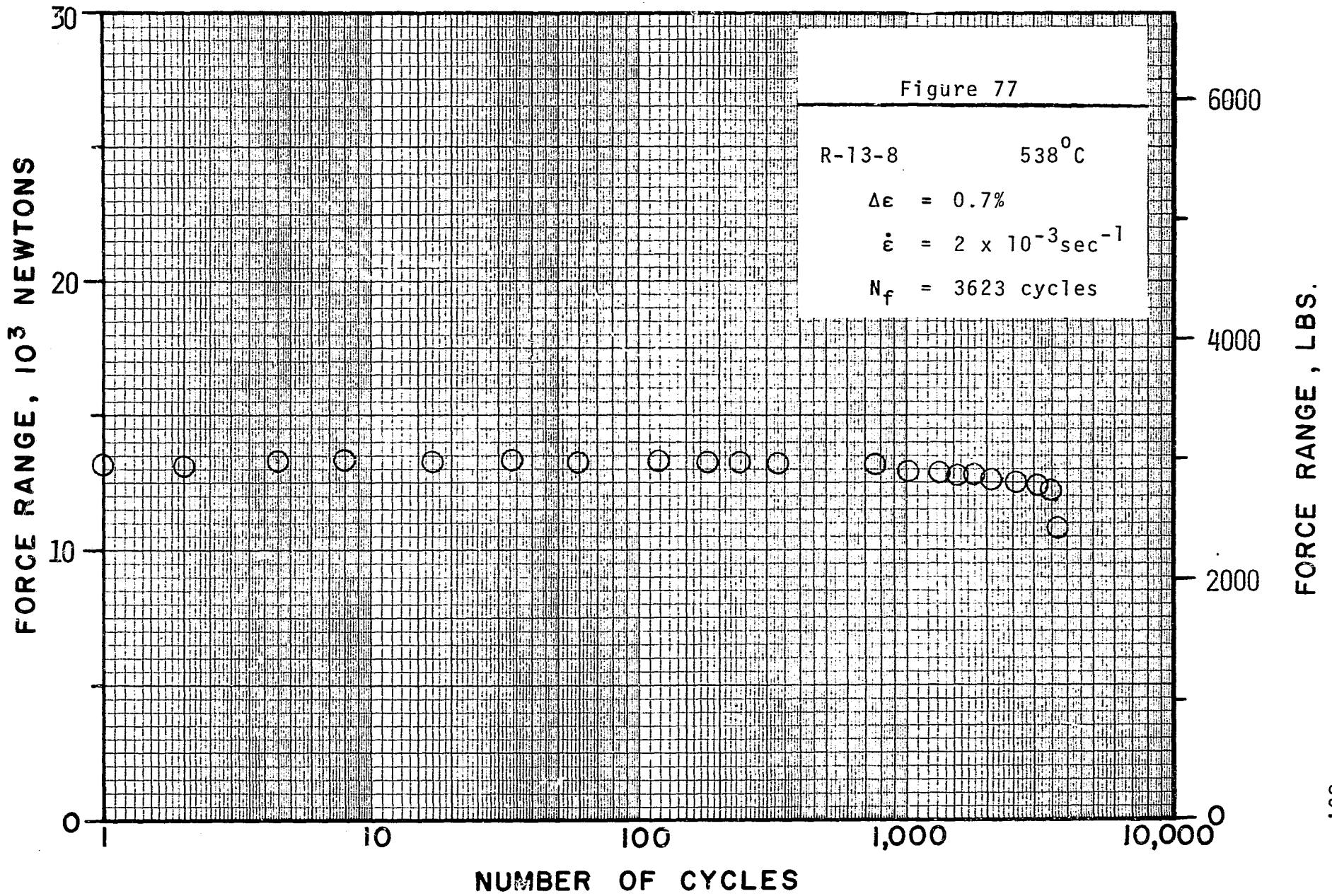
FORCE RANGE, LBS.

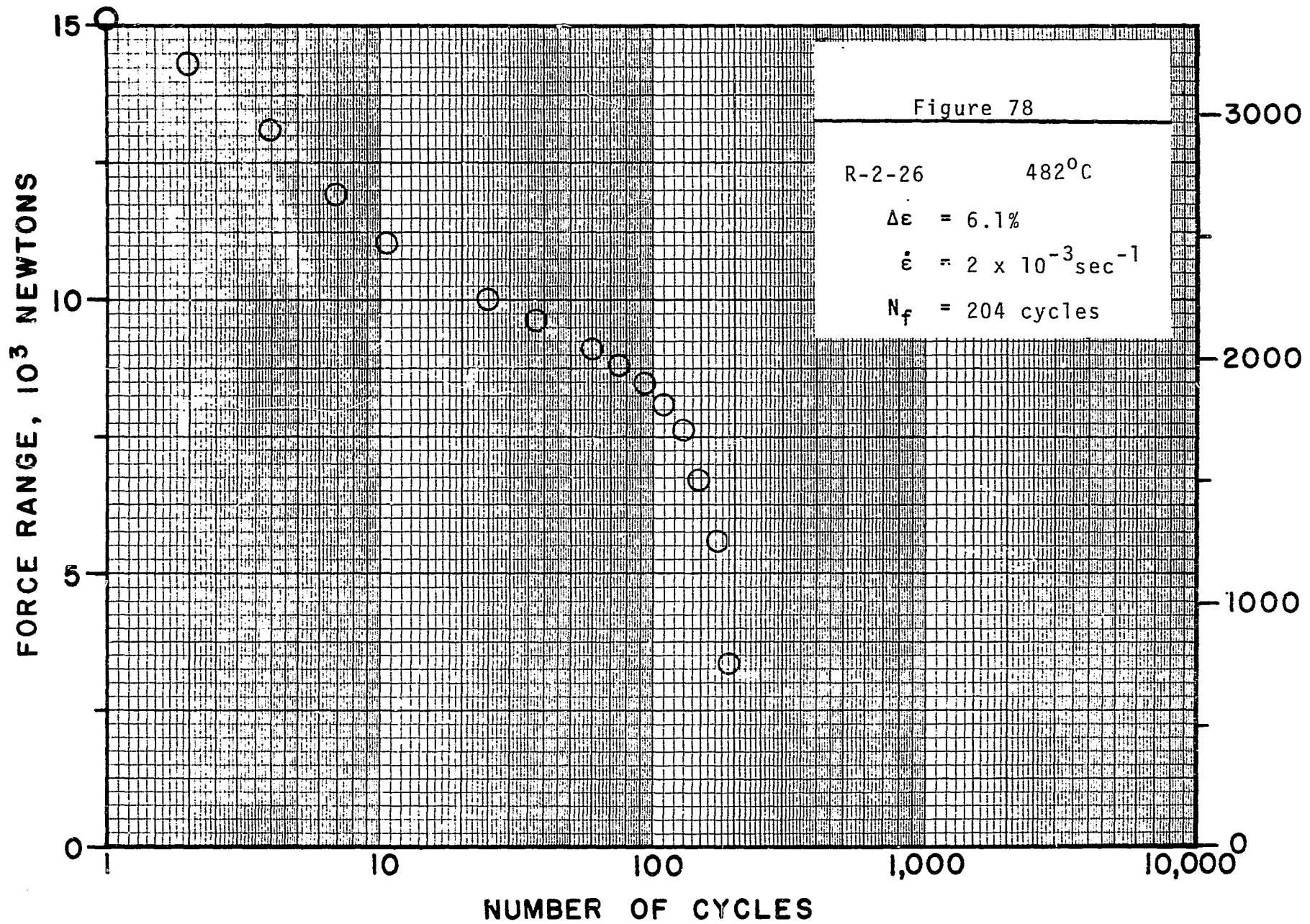


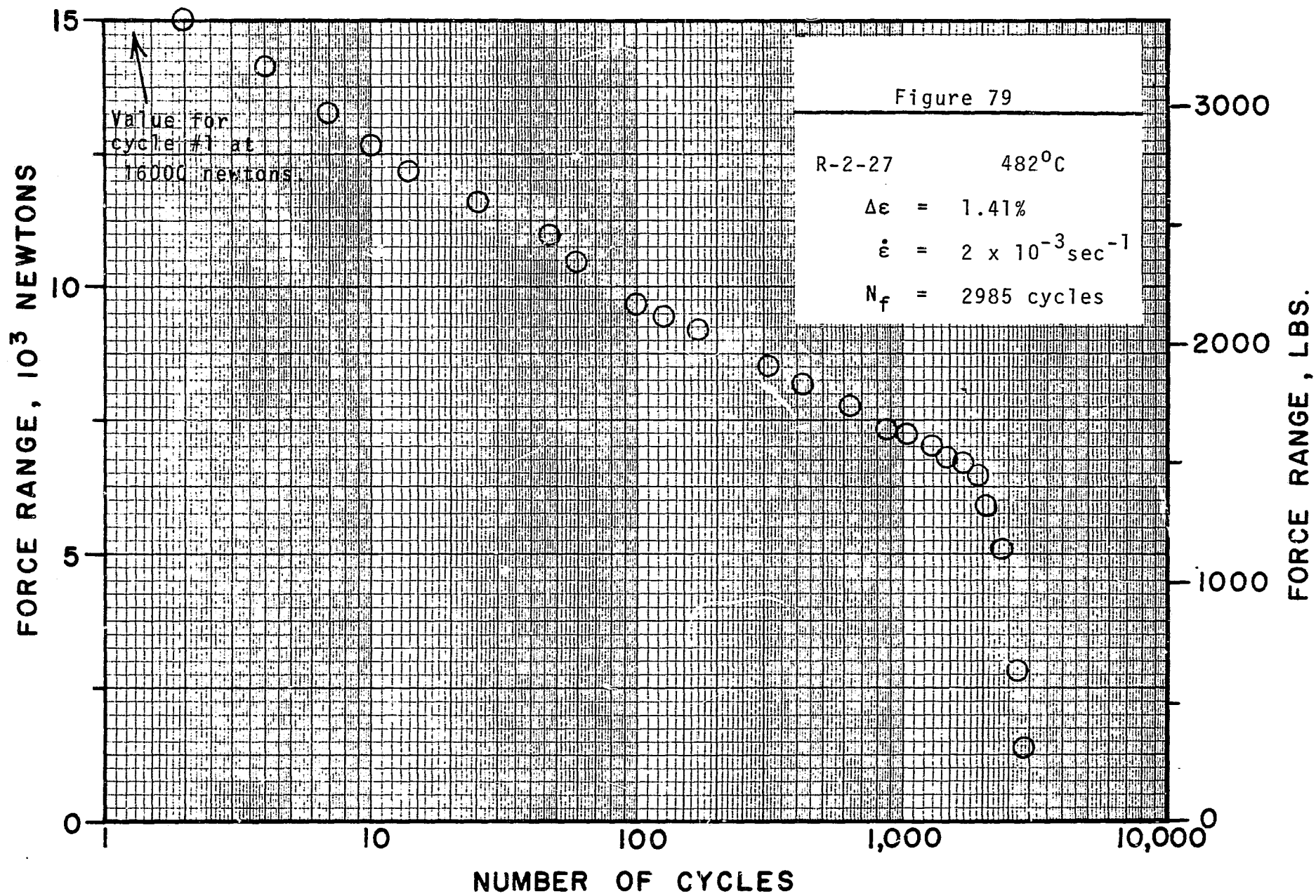


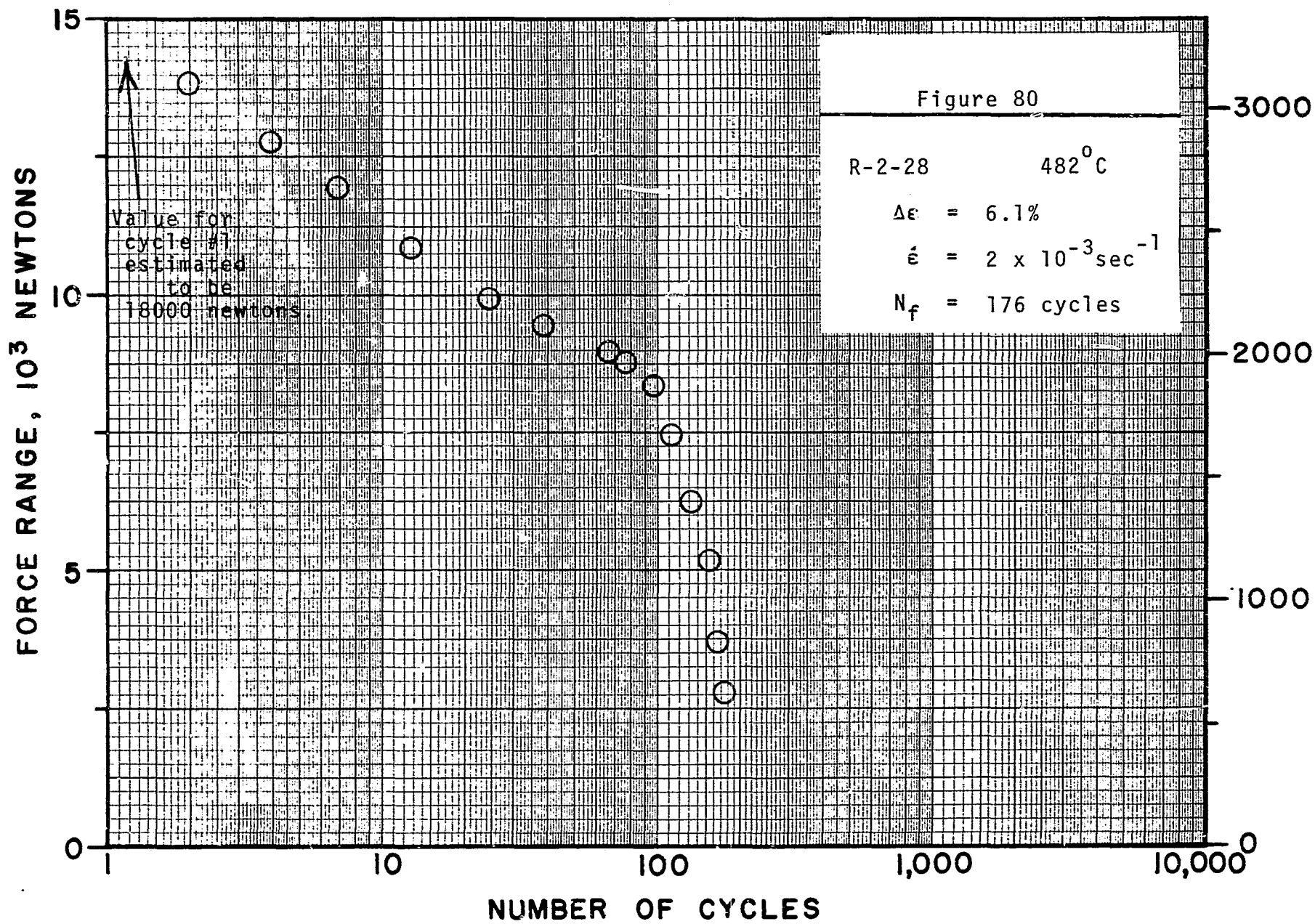




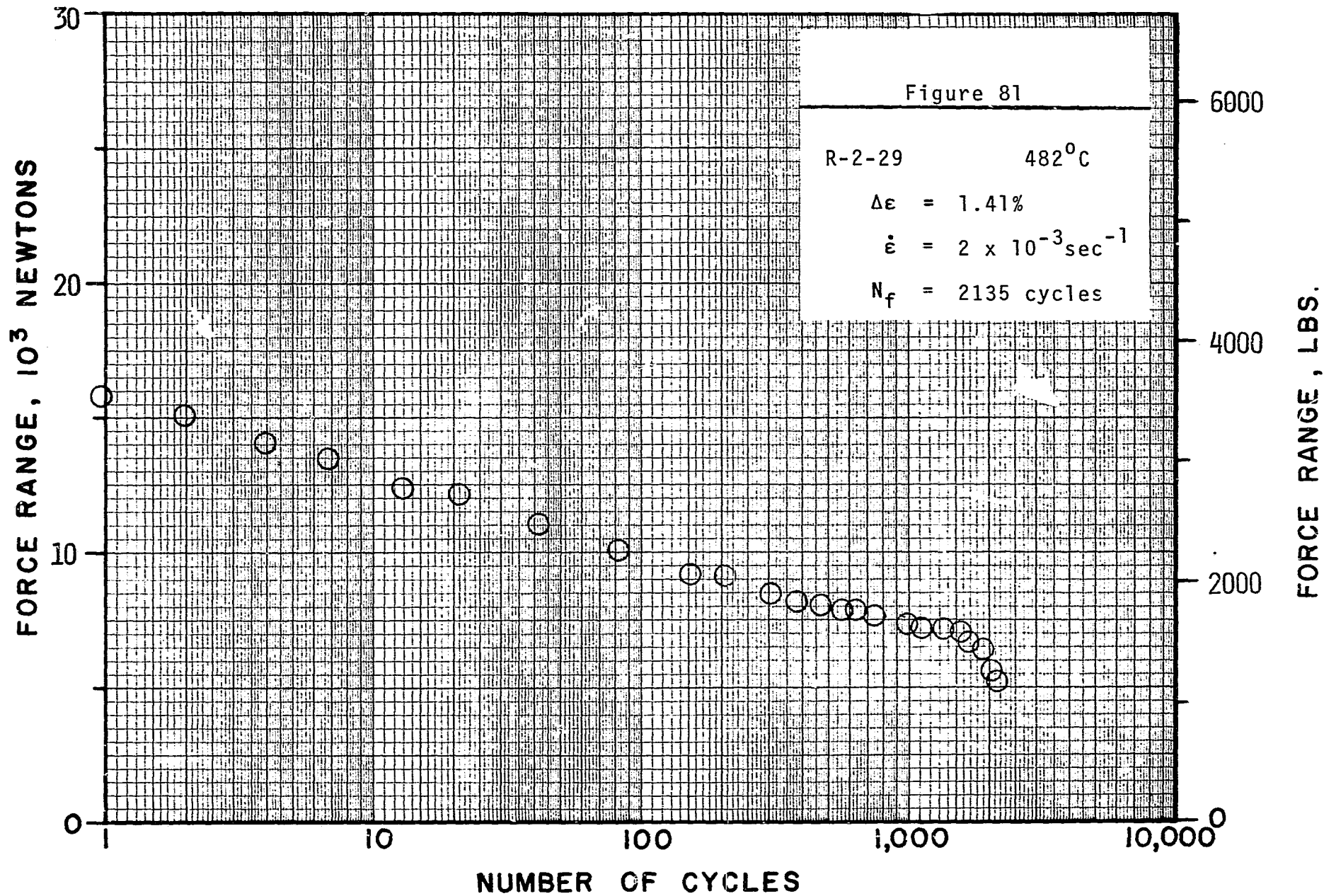


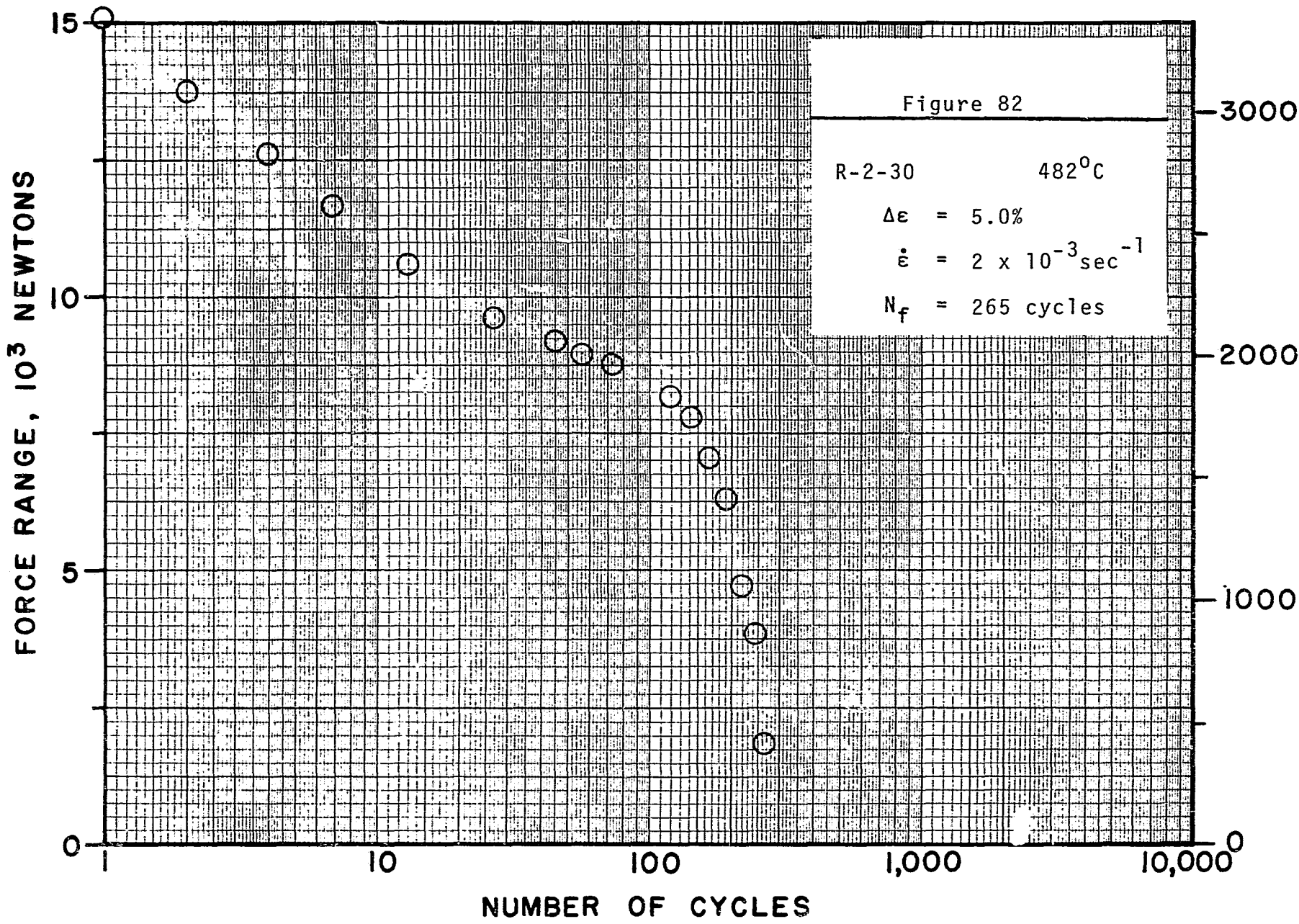




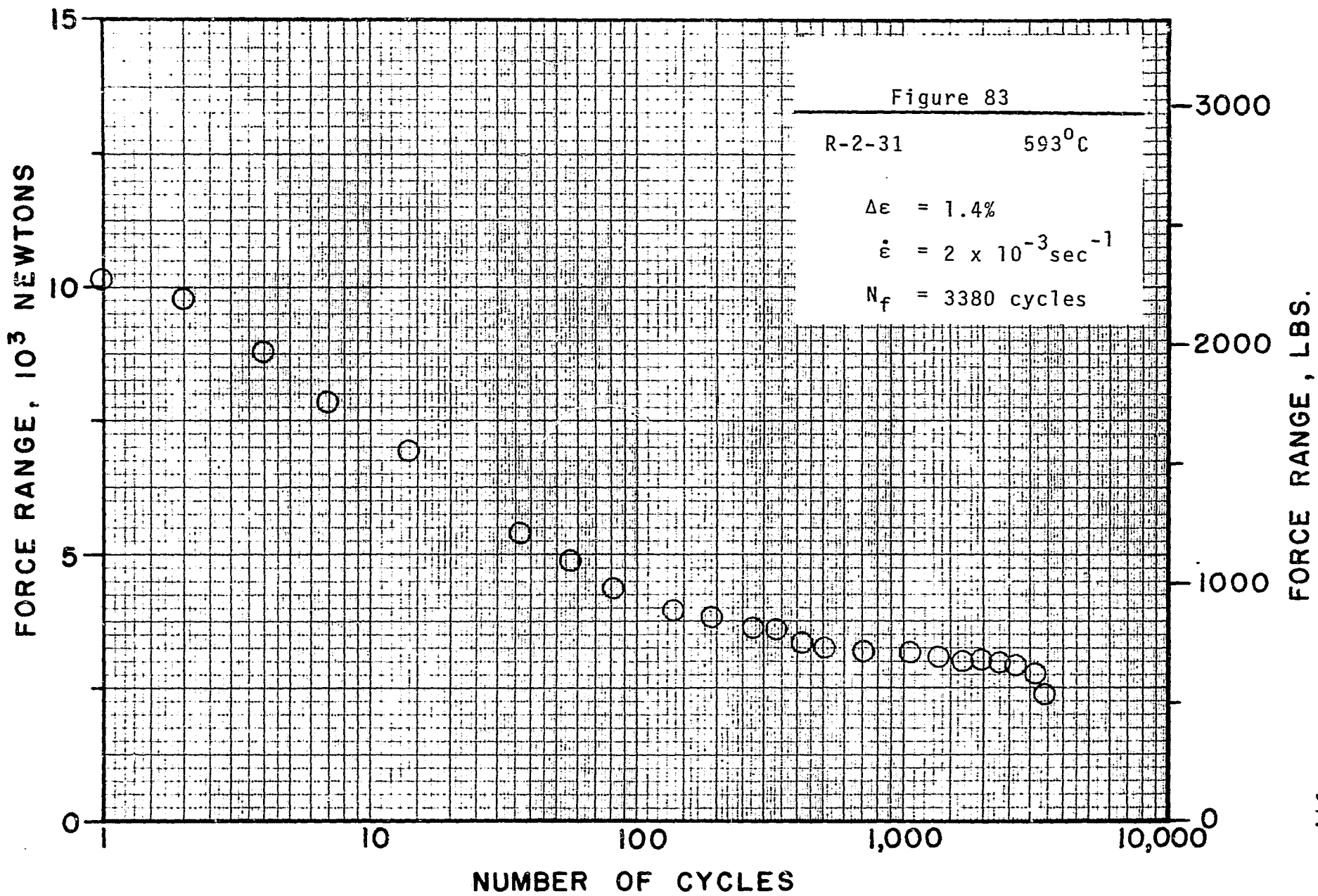


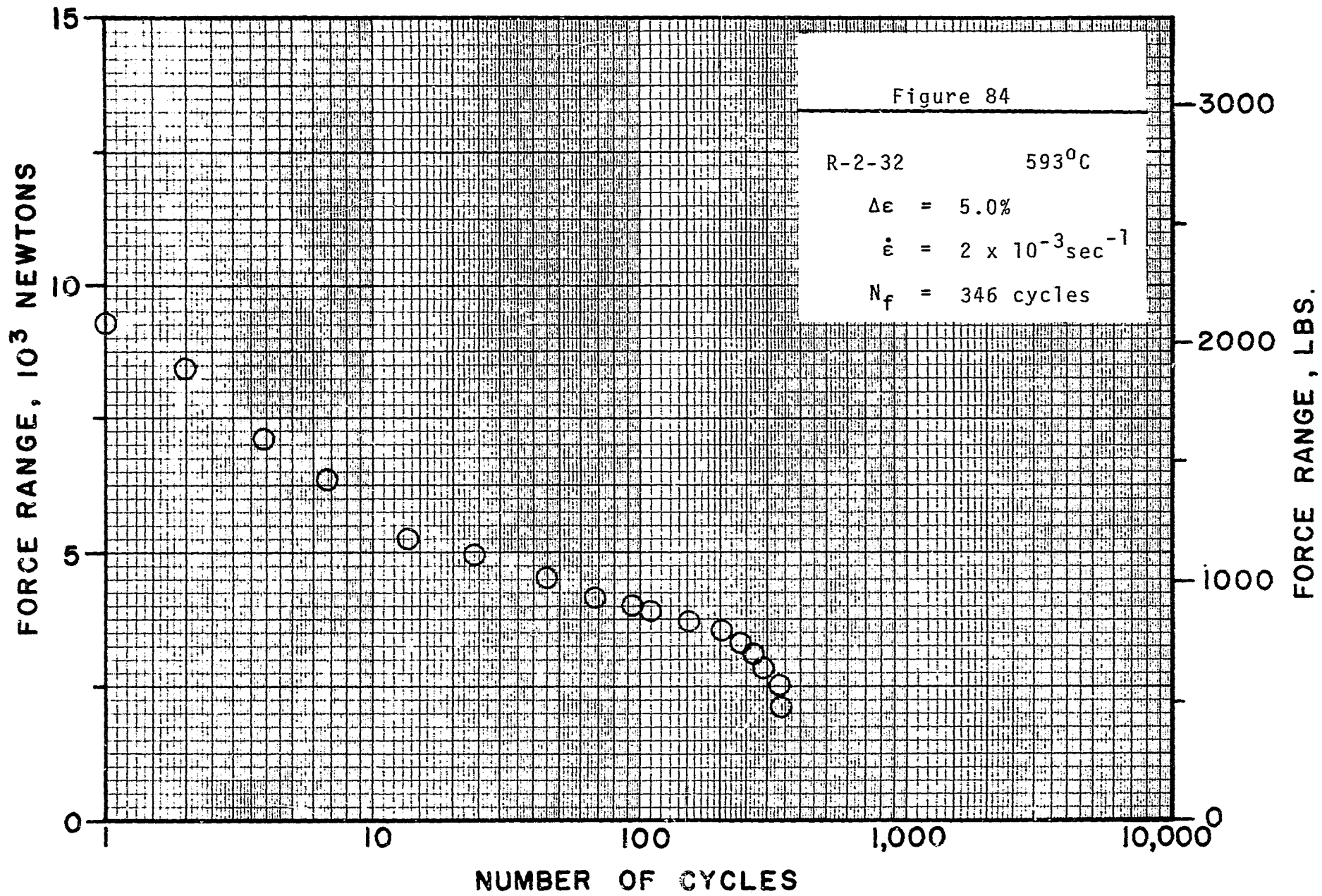
FORCE RANGE, LBS.

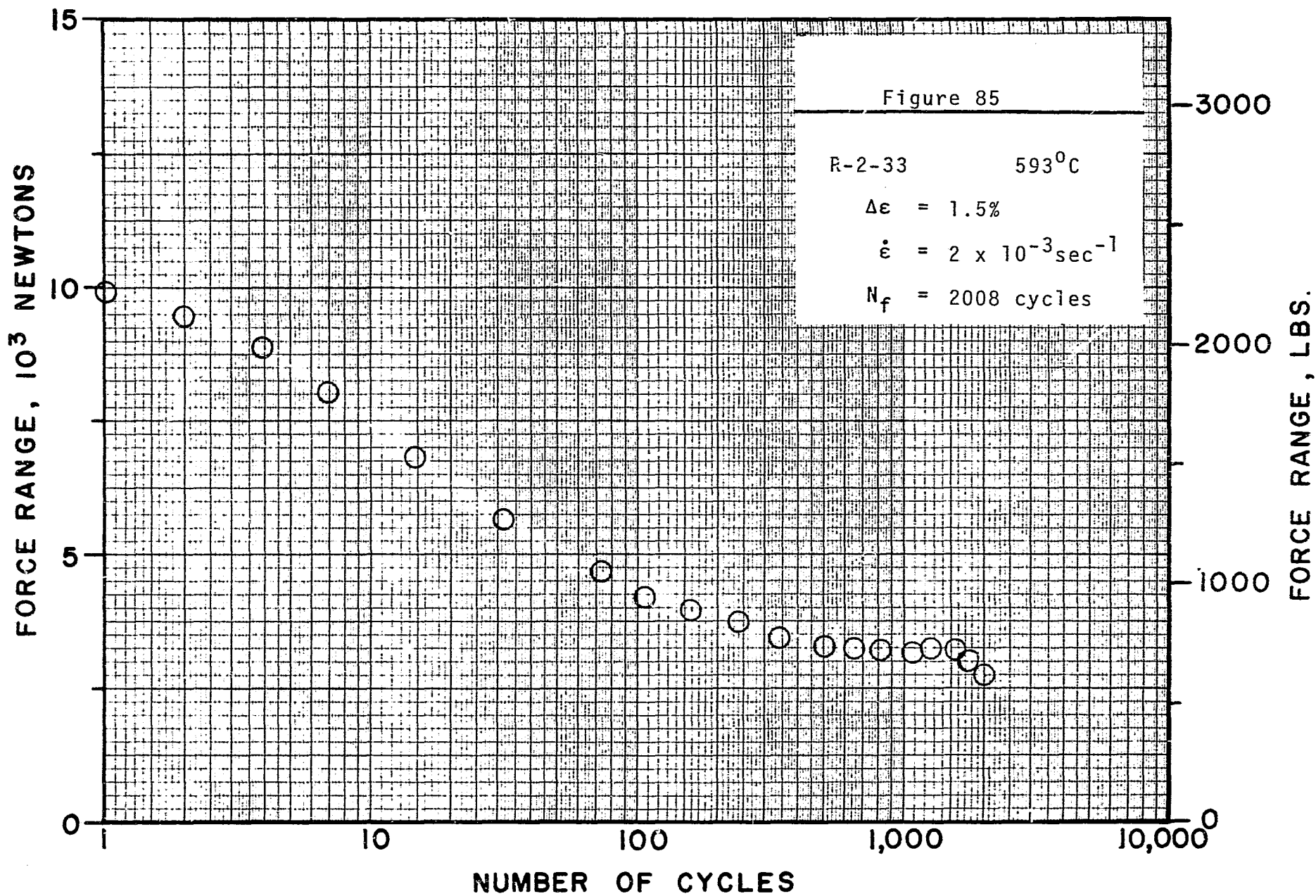


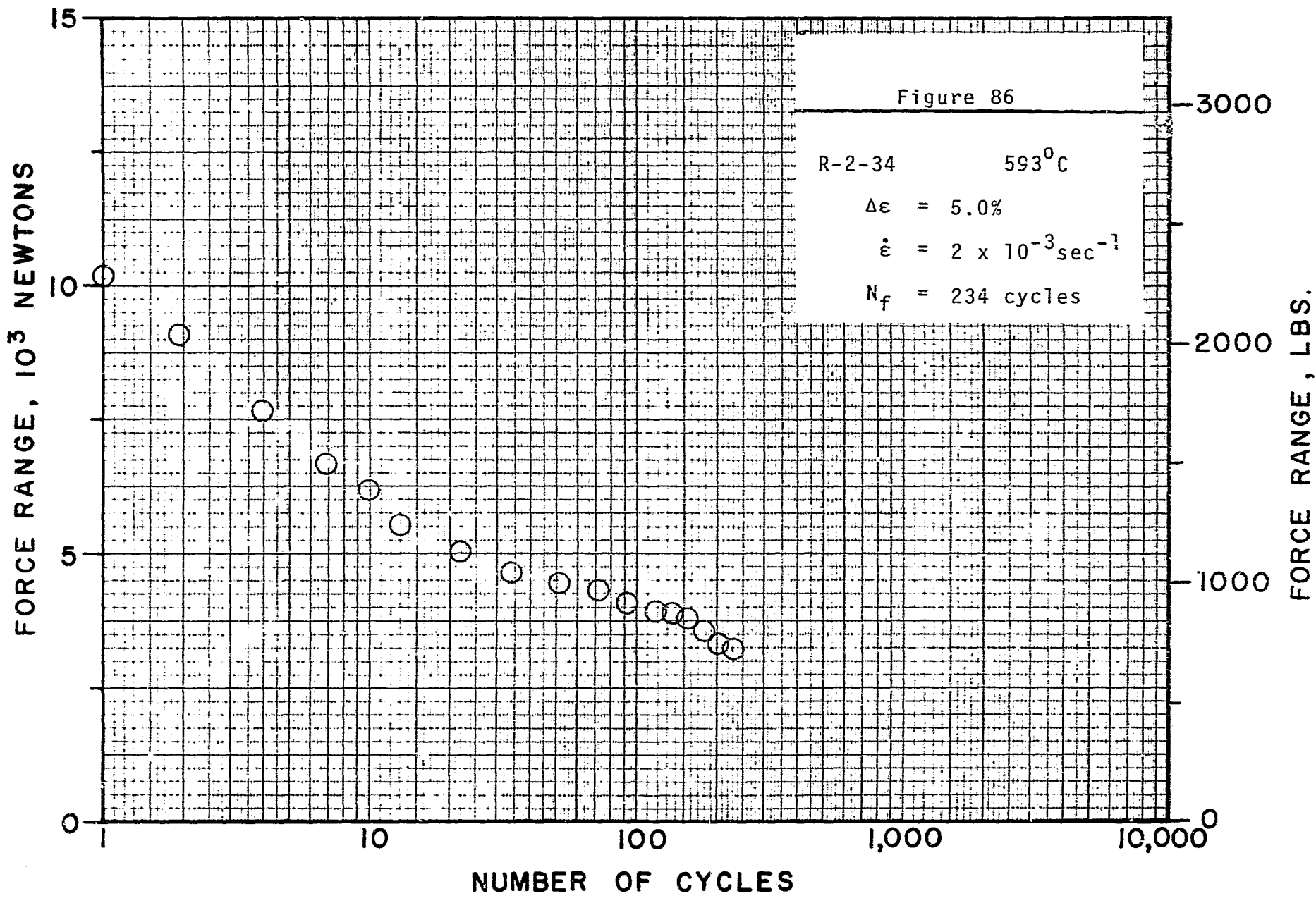


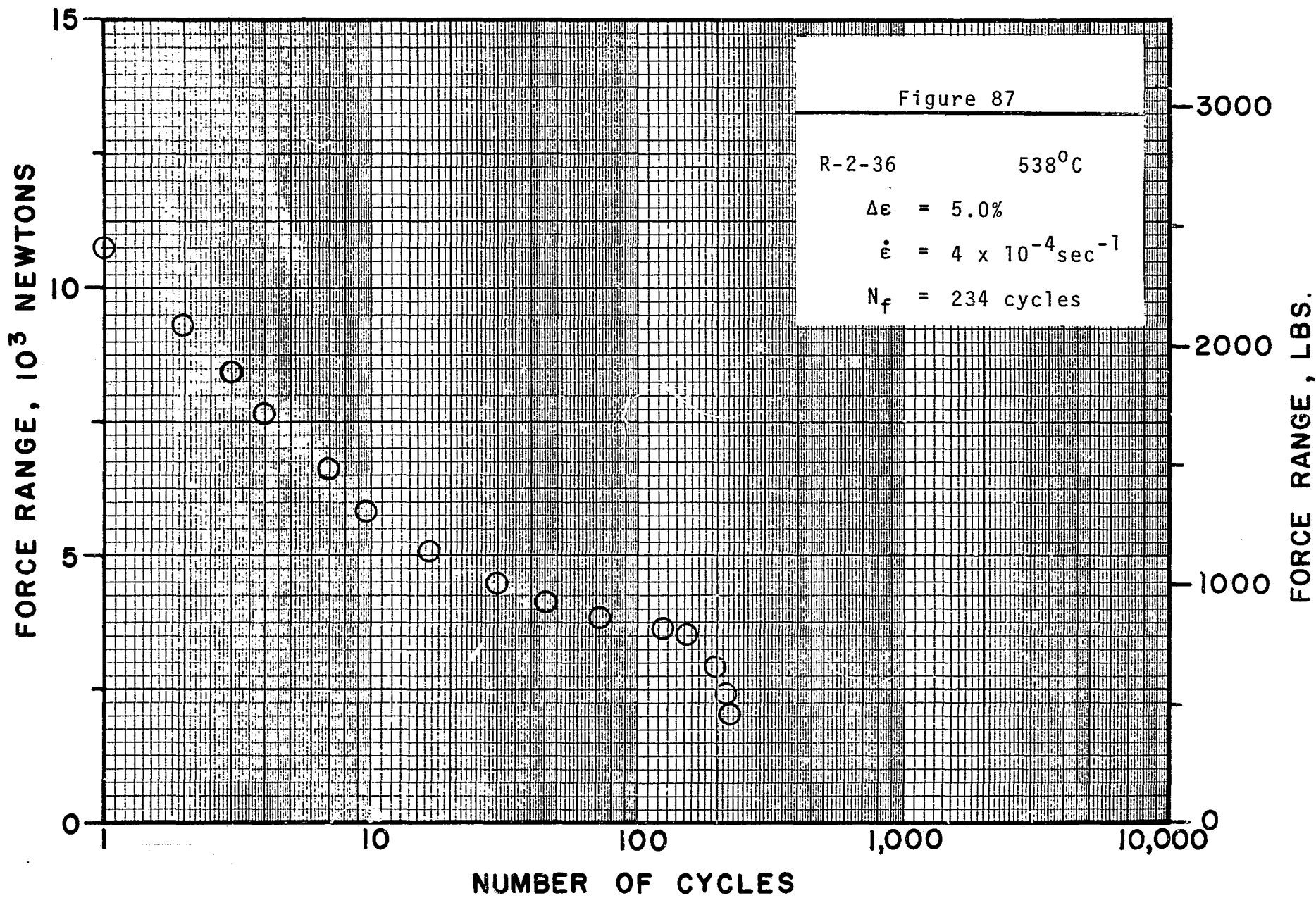
FORCE RANGE, LBS.

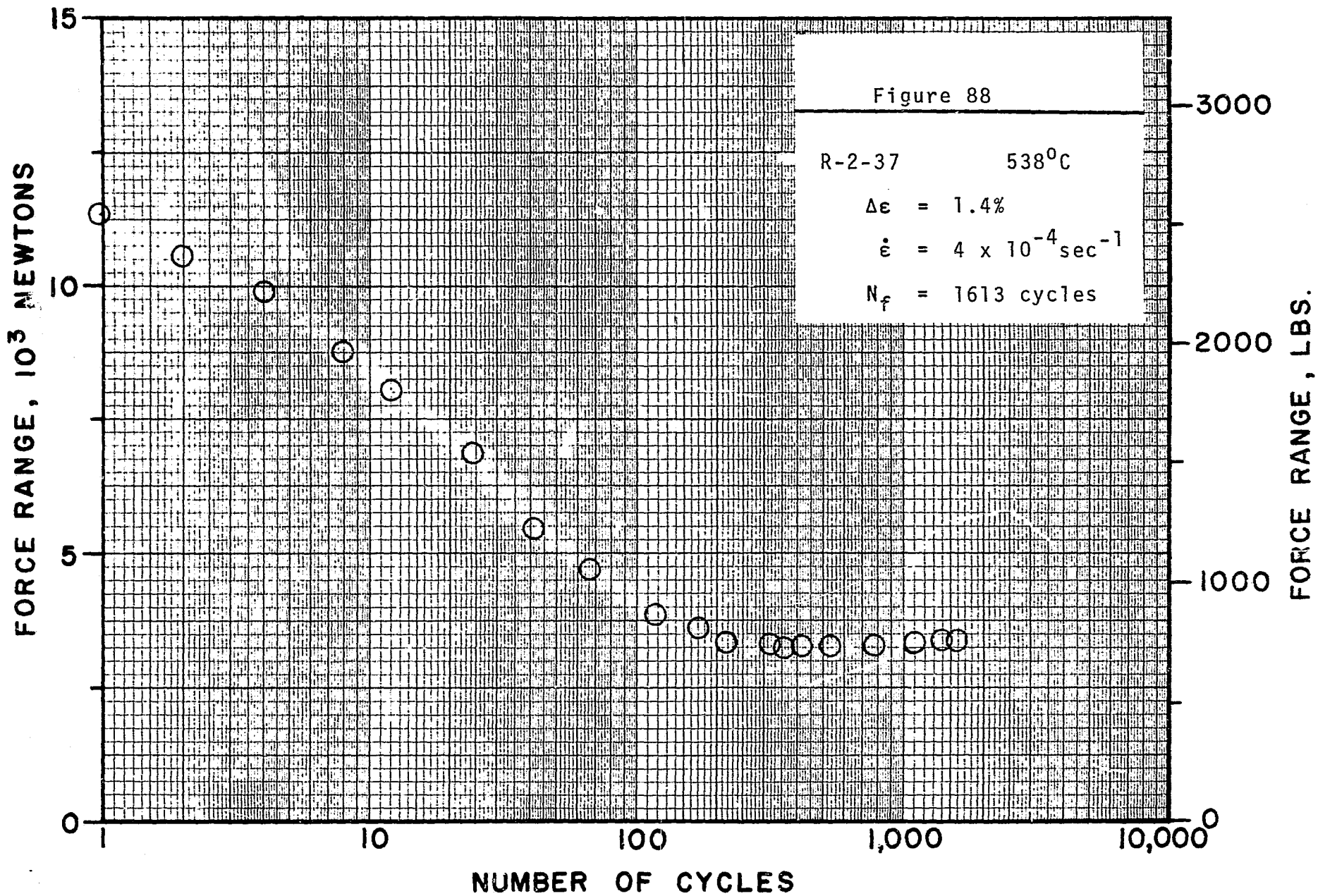


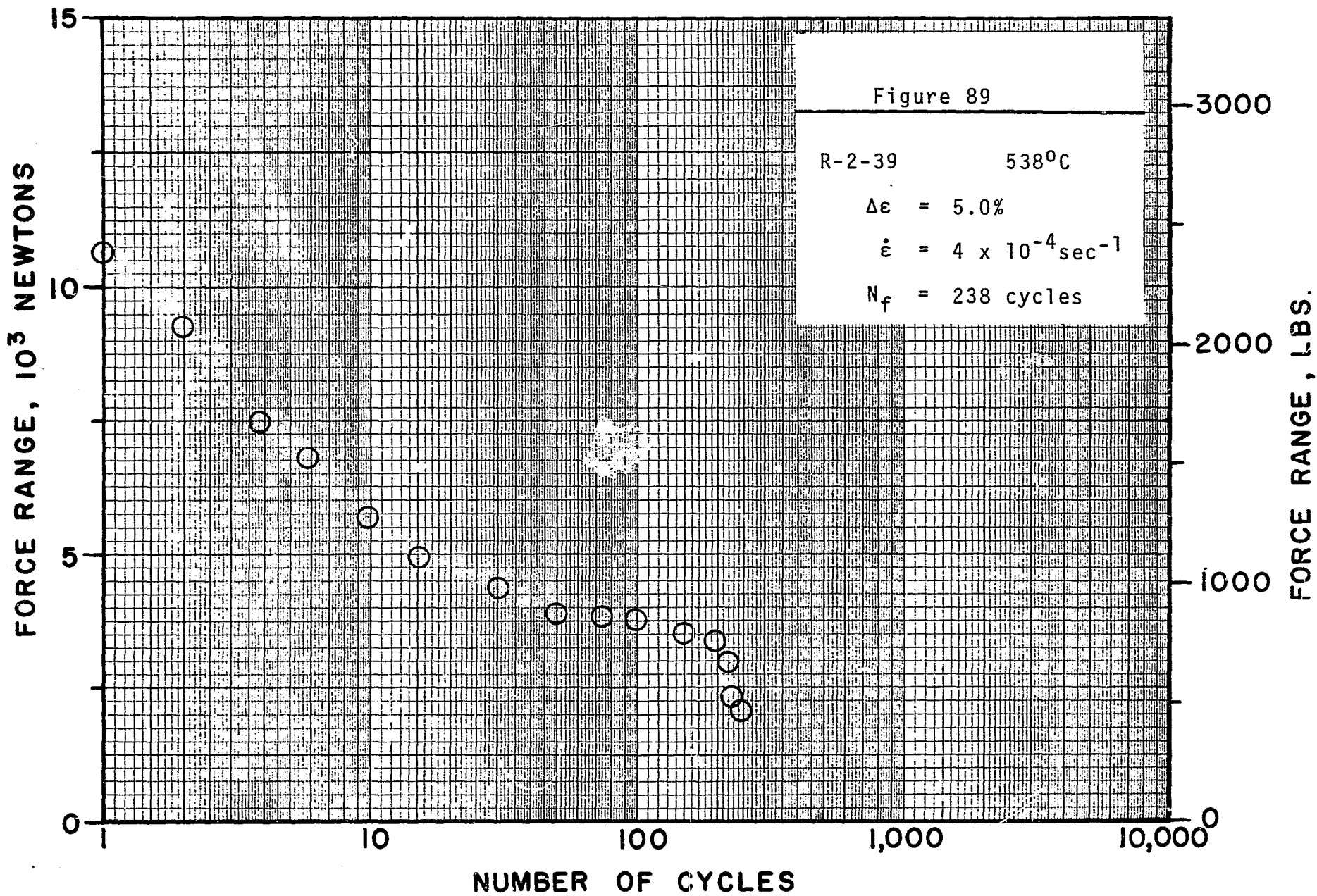


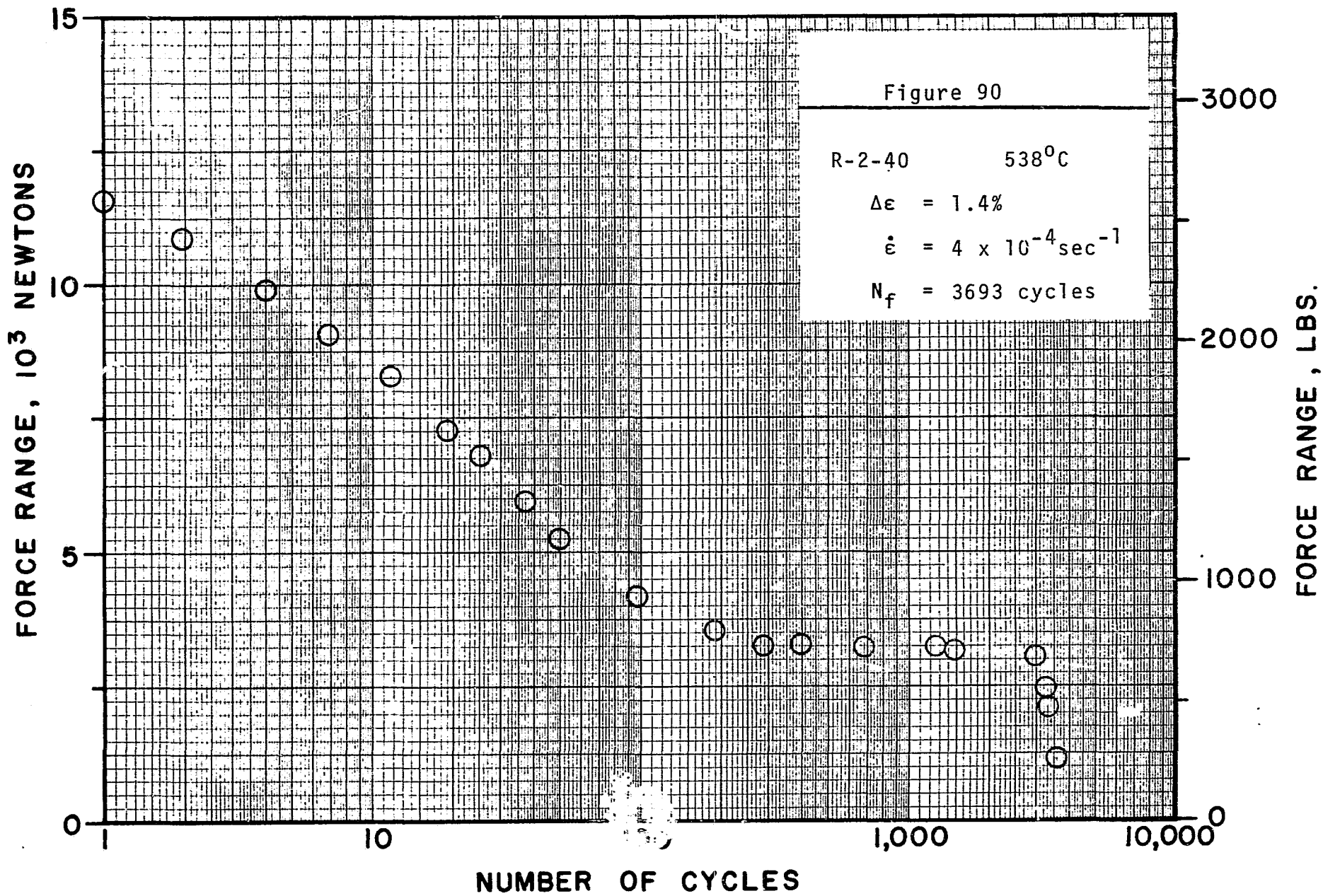


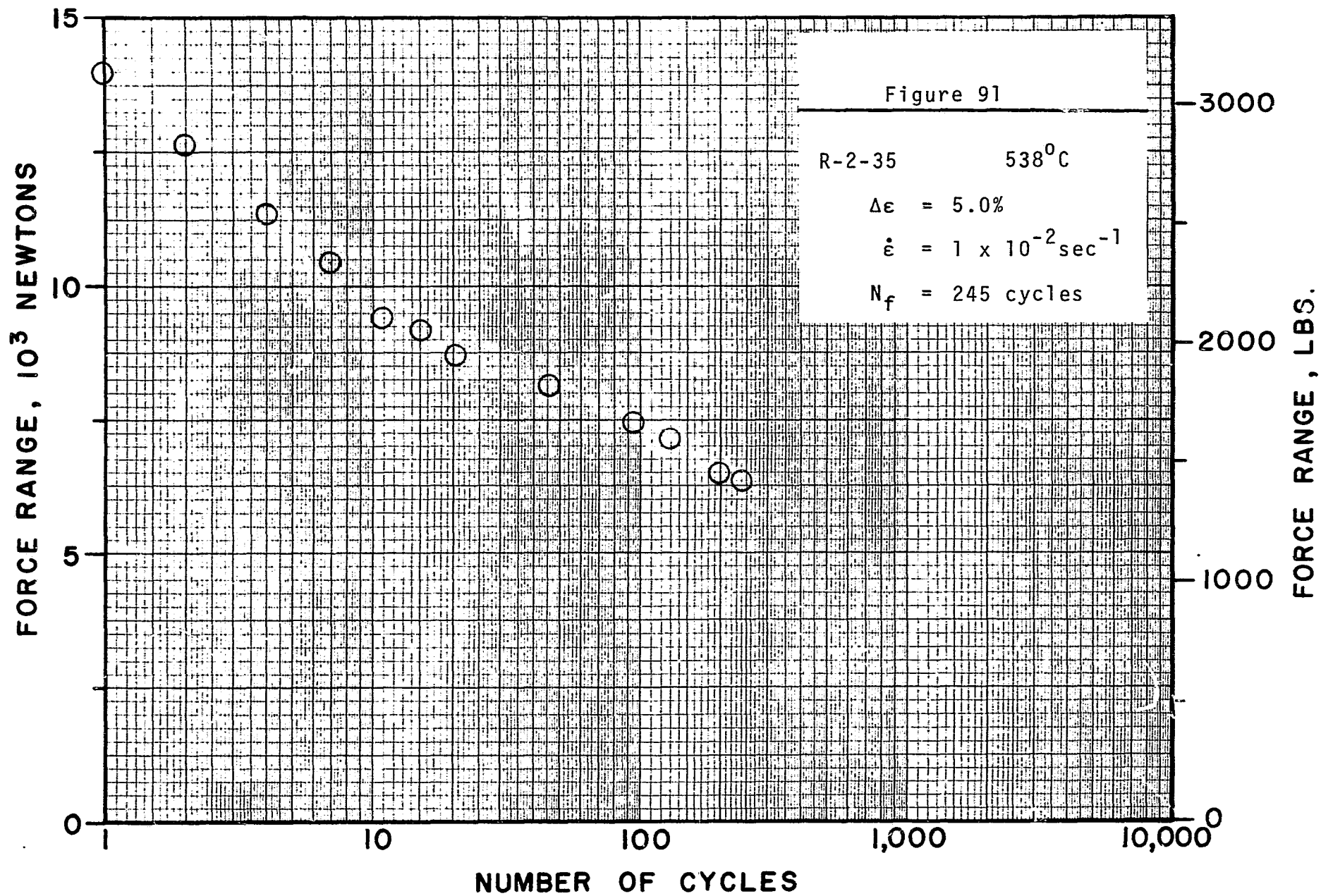


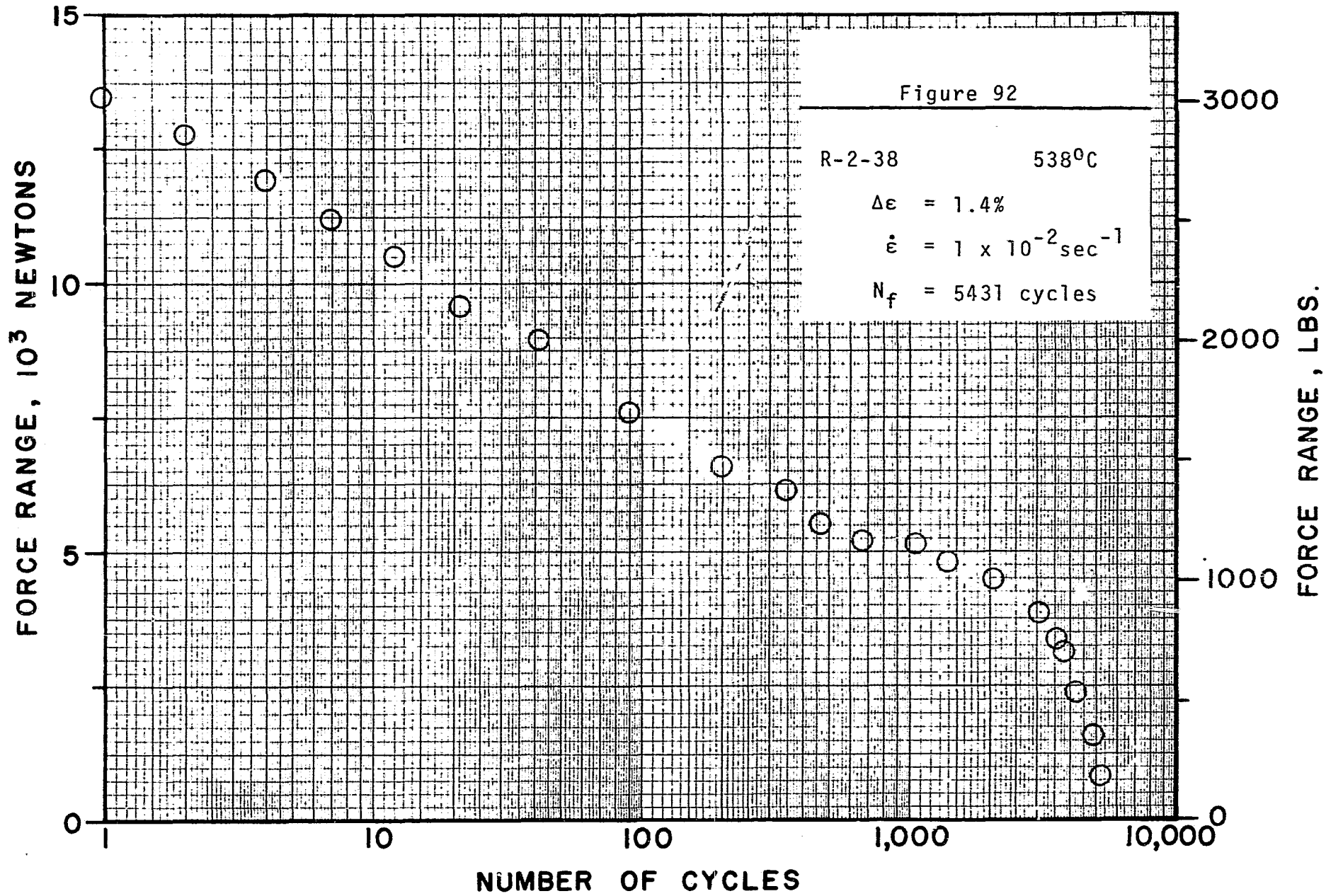


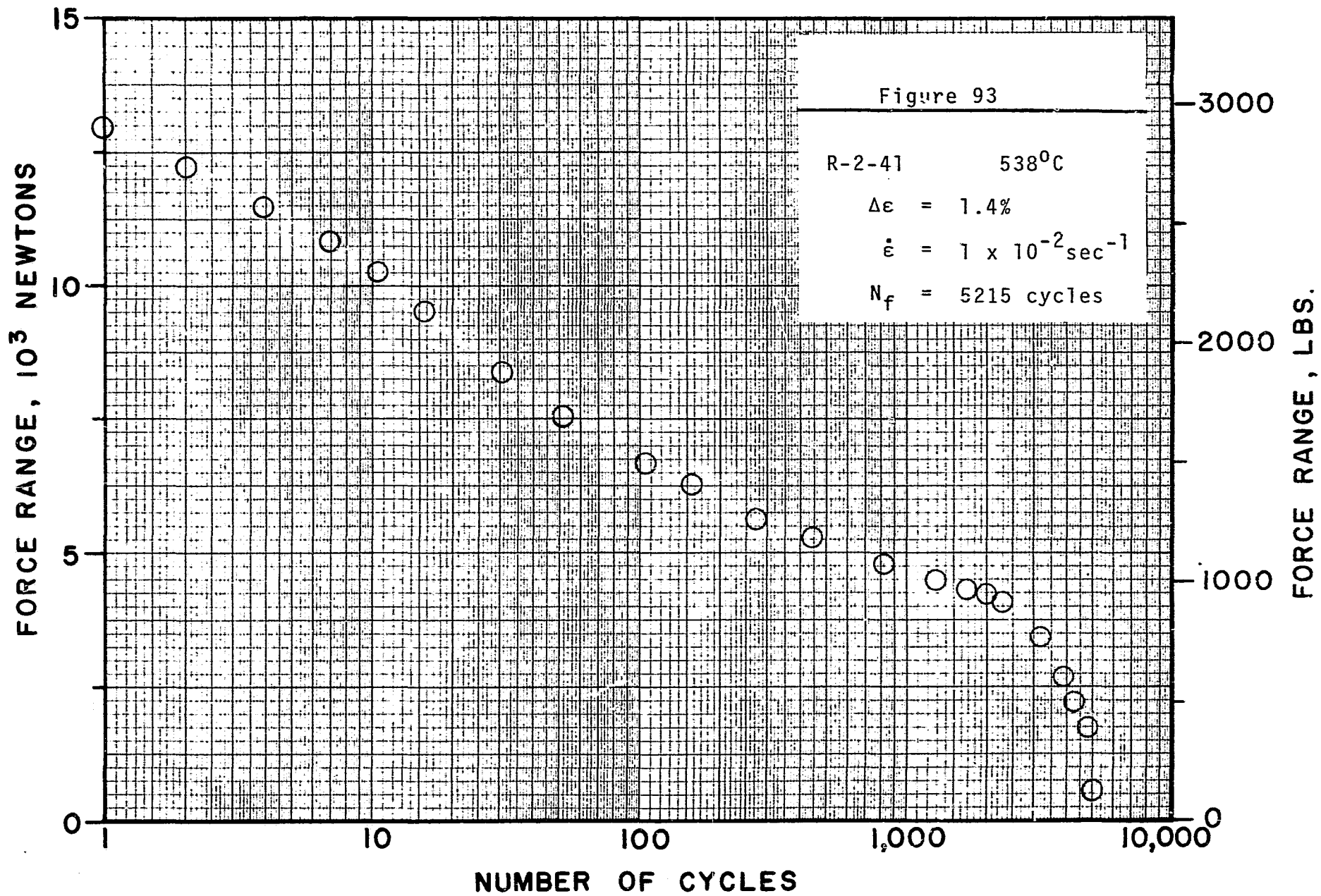


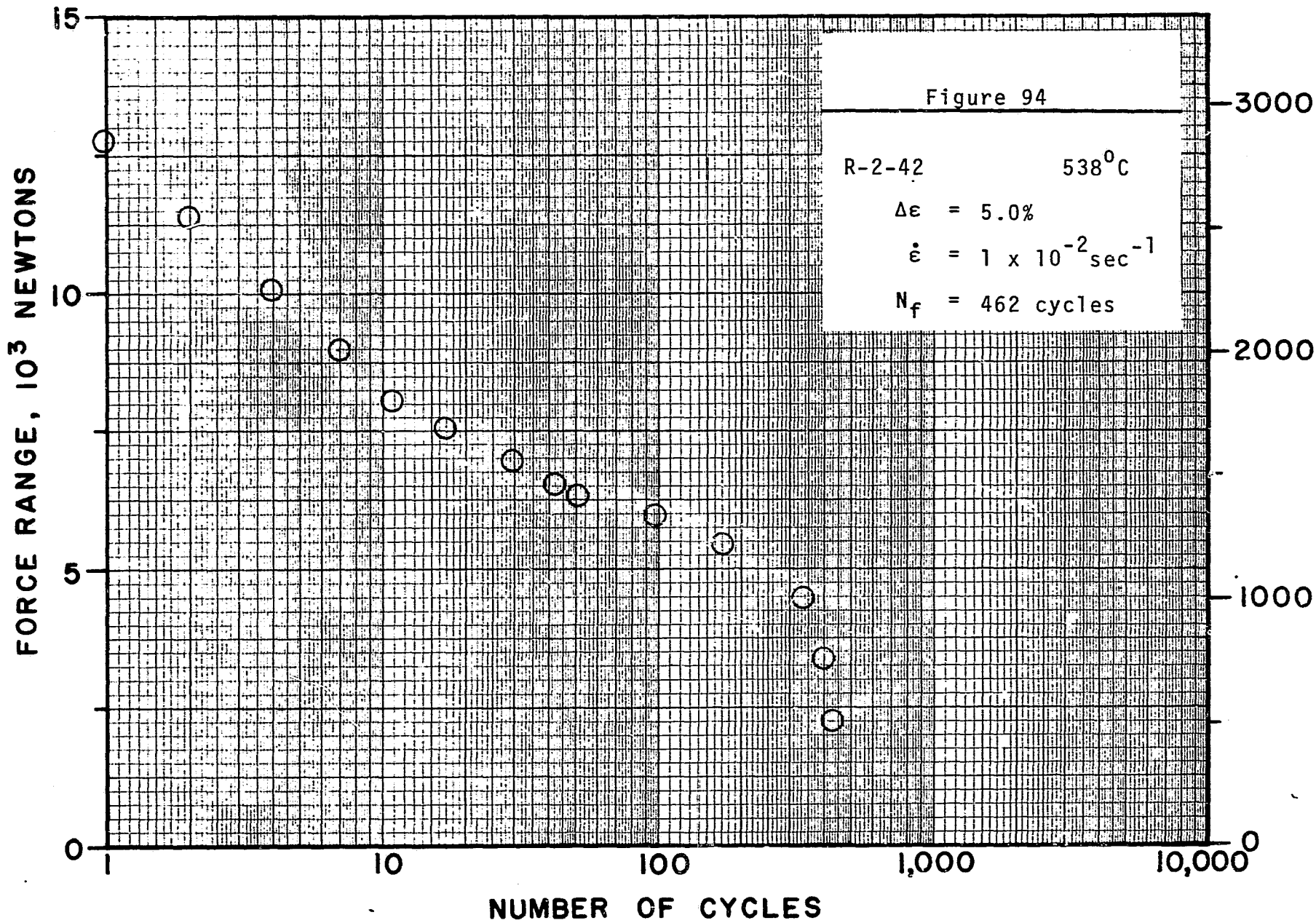


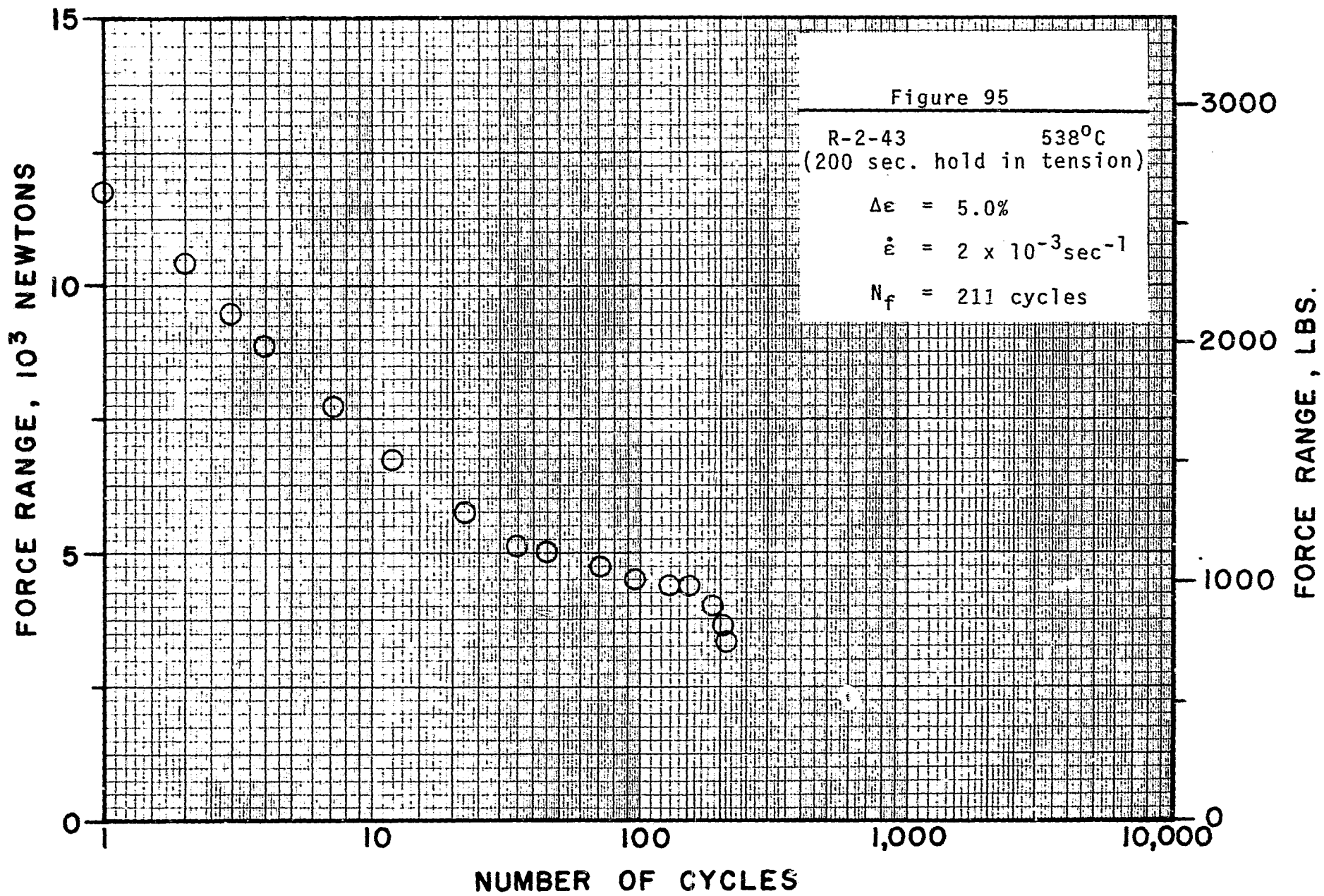


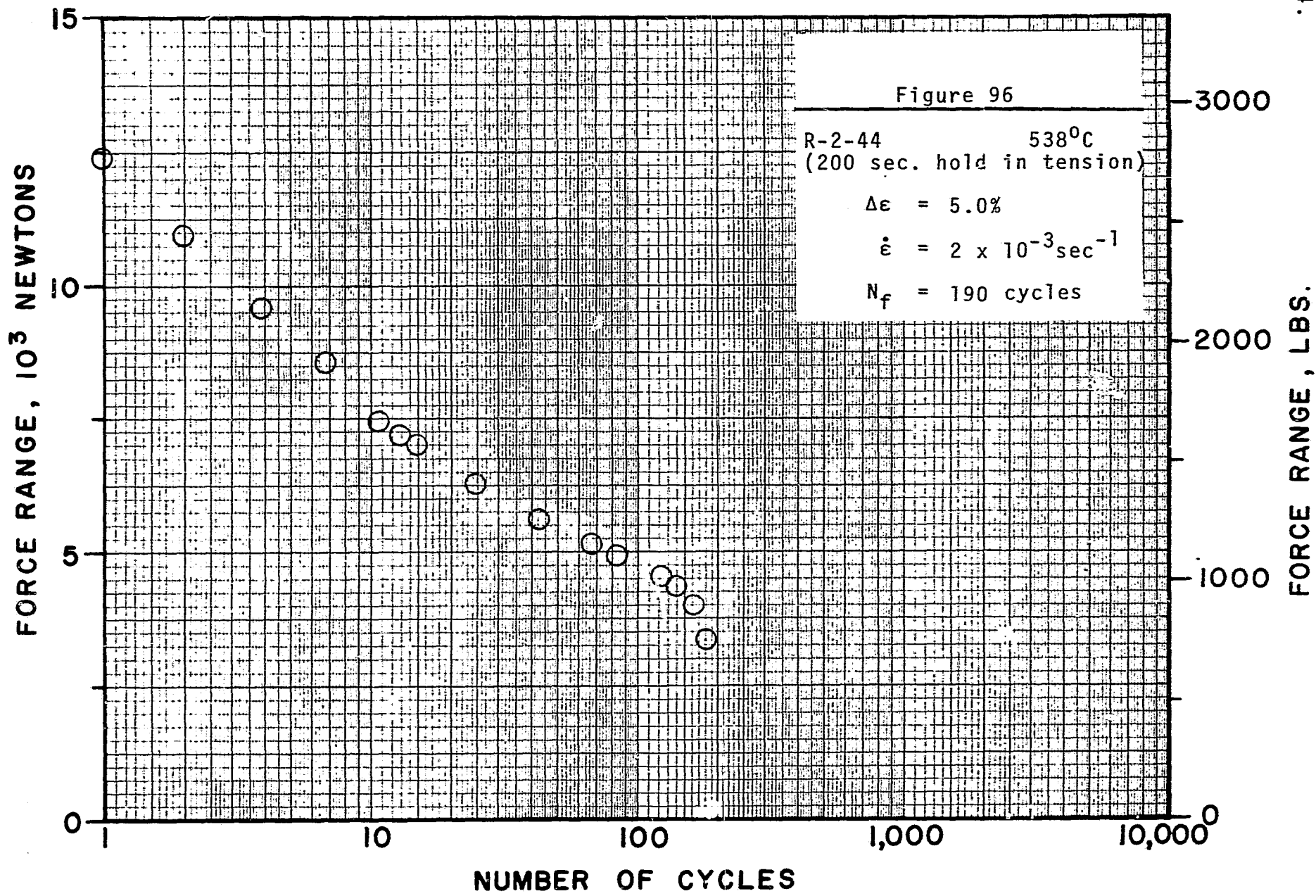


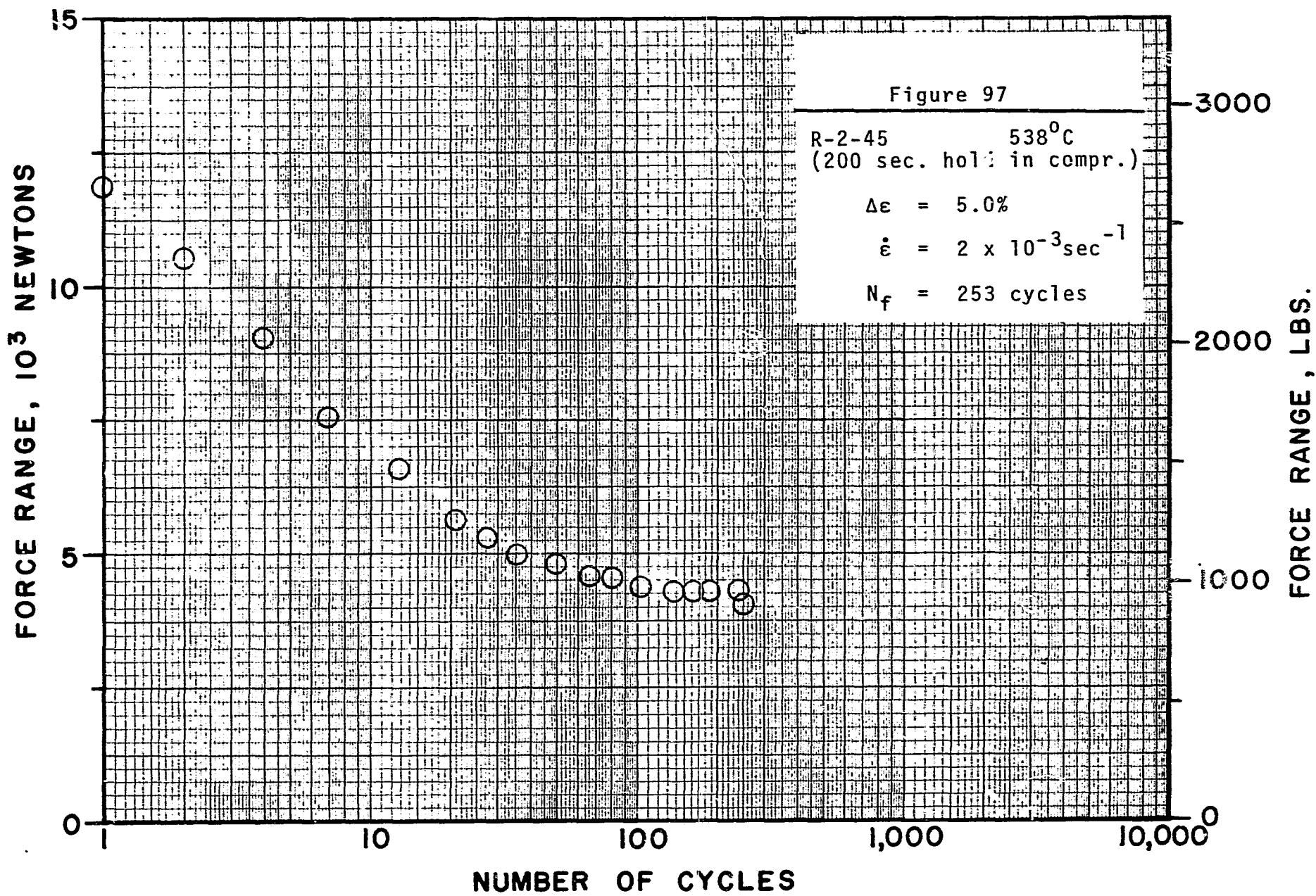


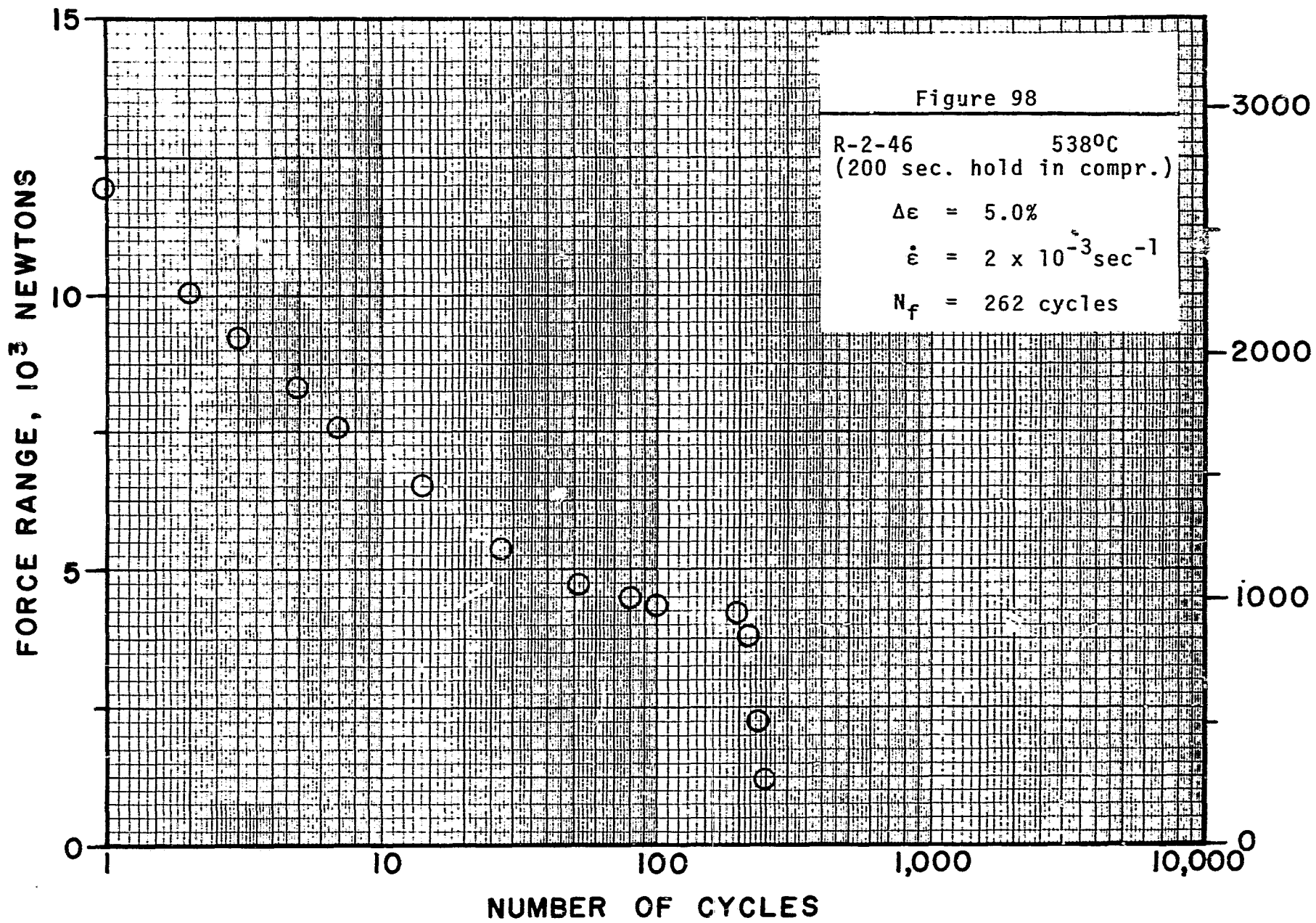


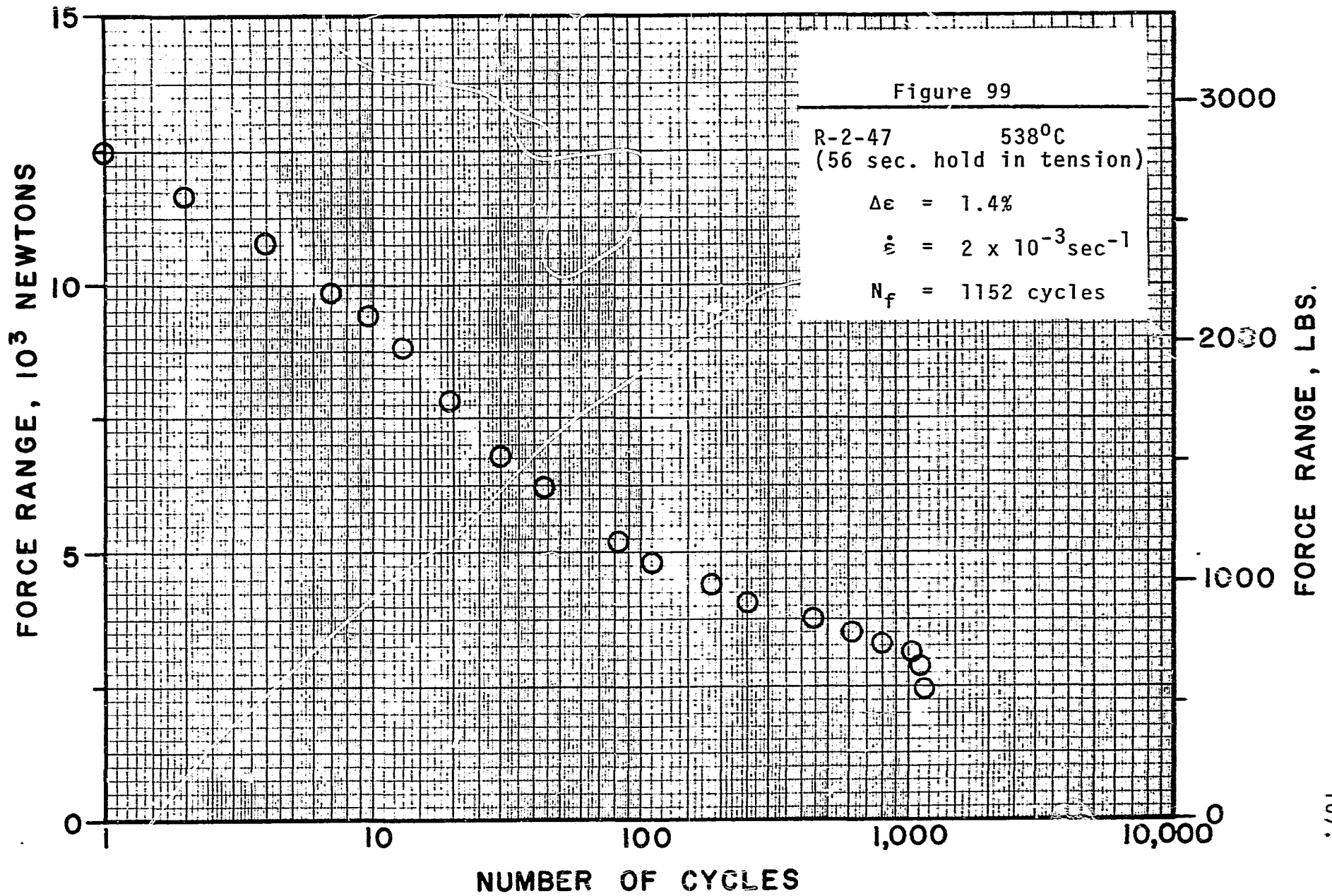


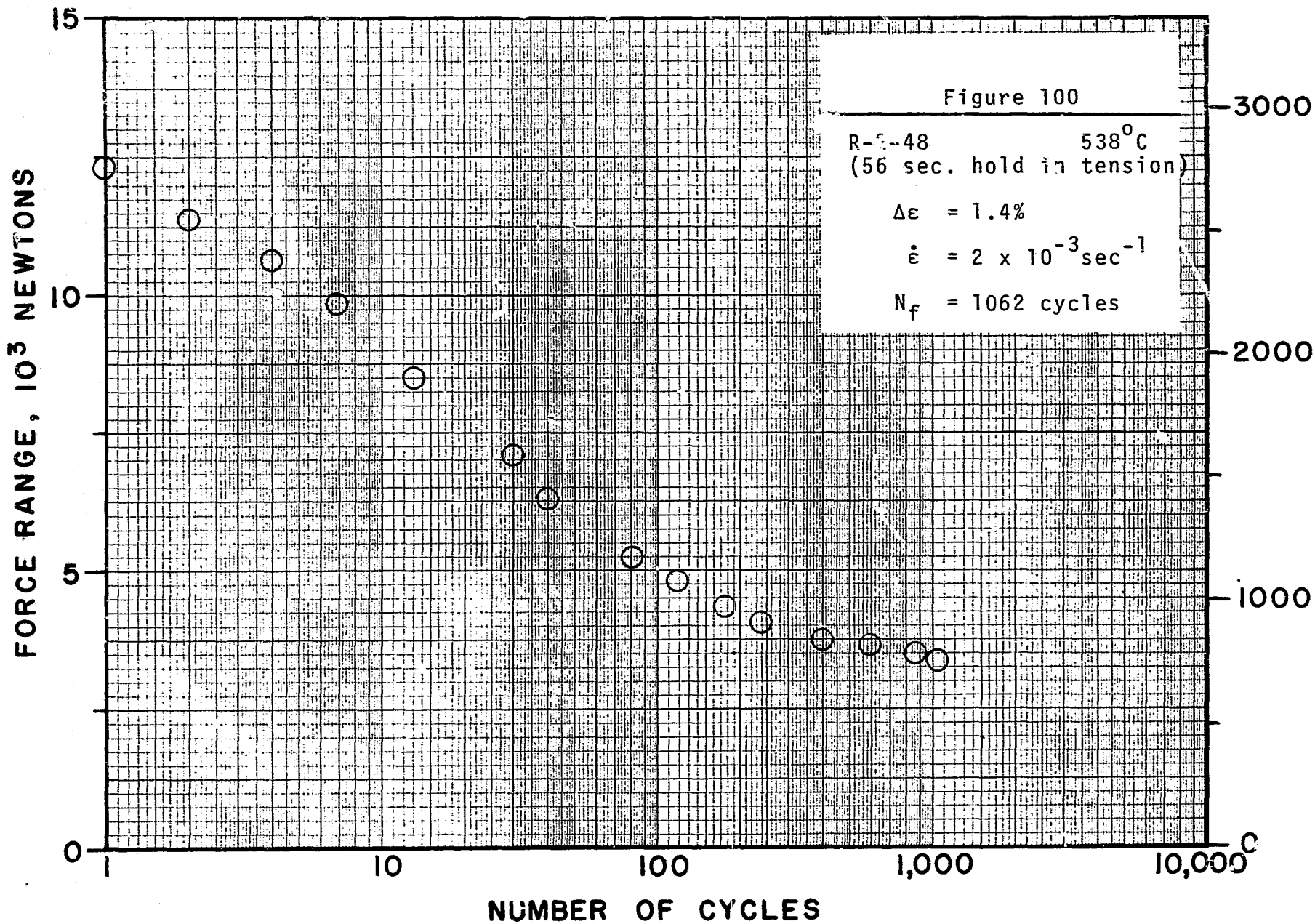




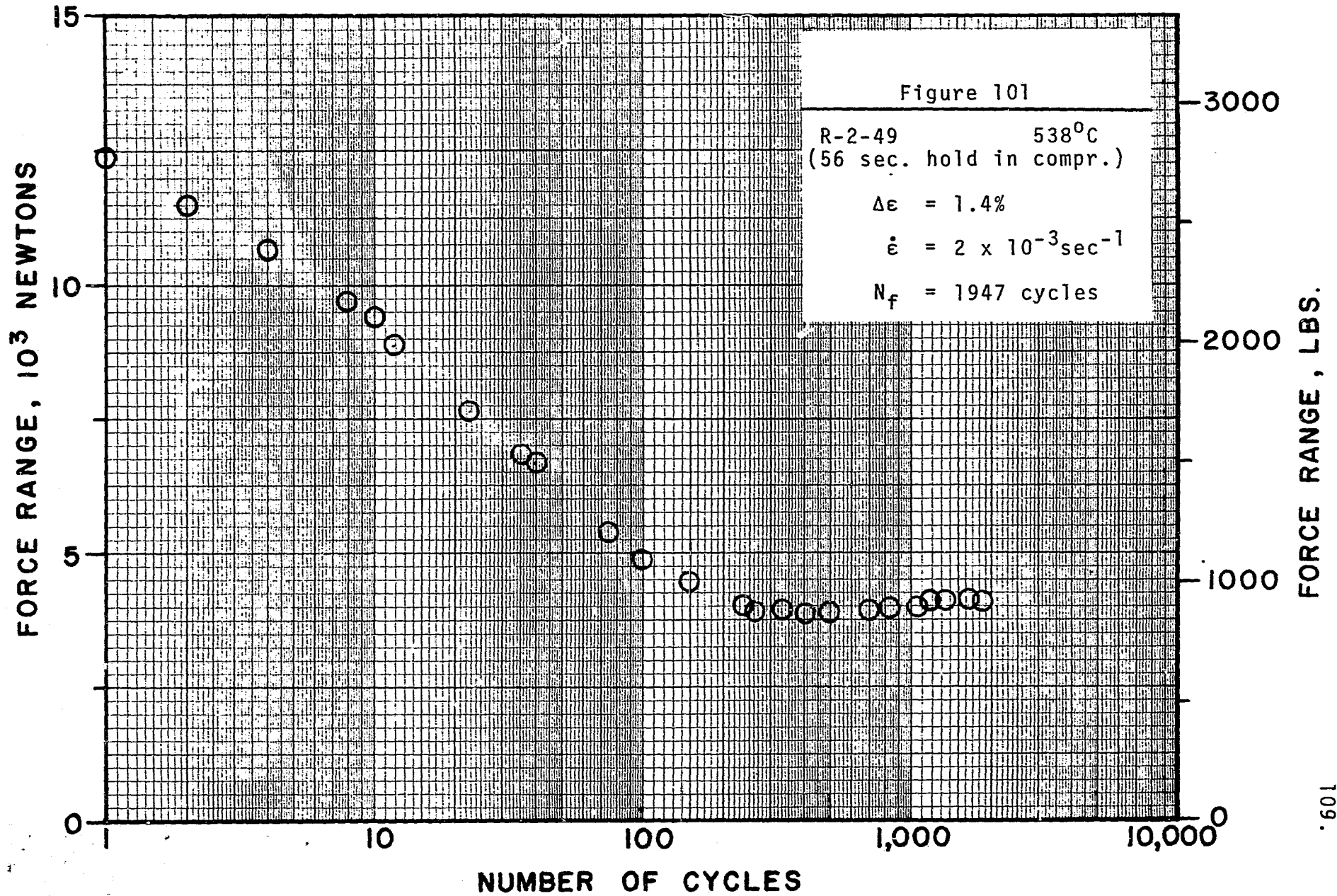


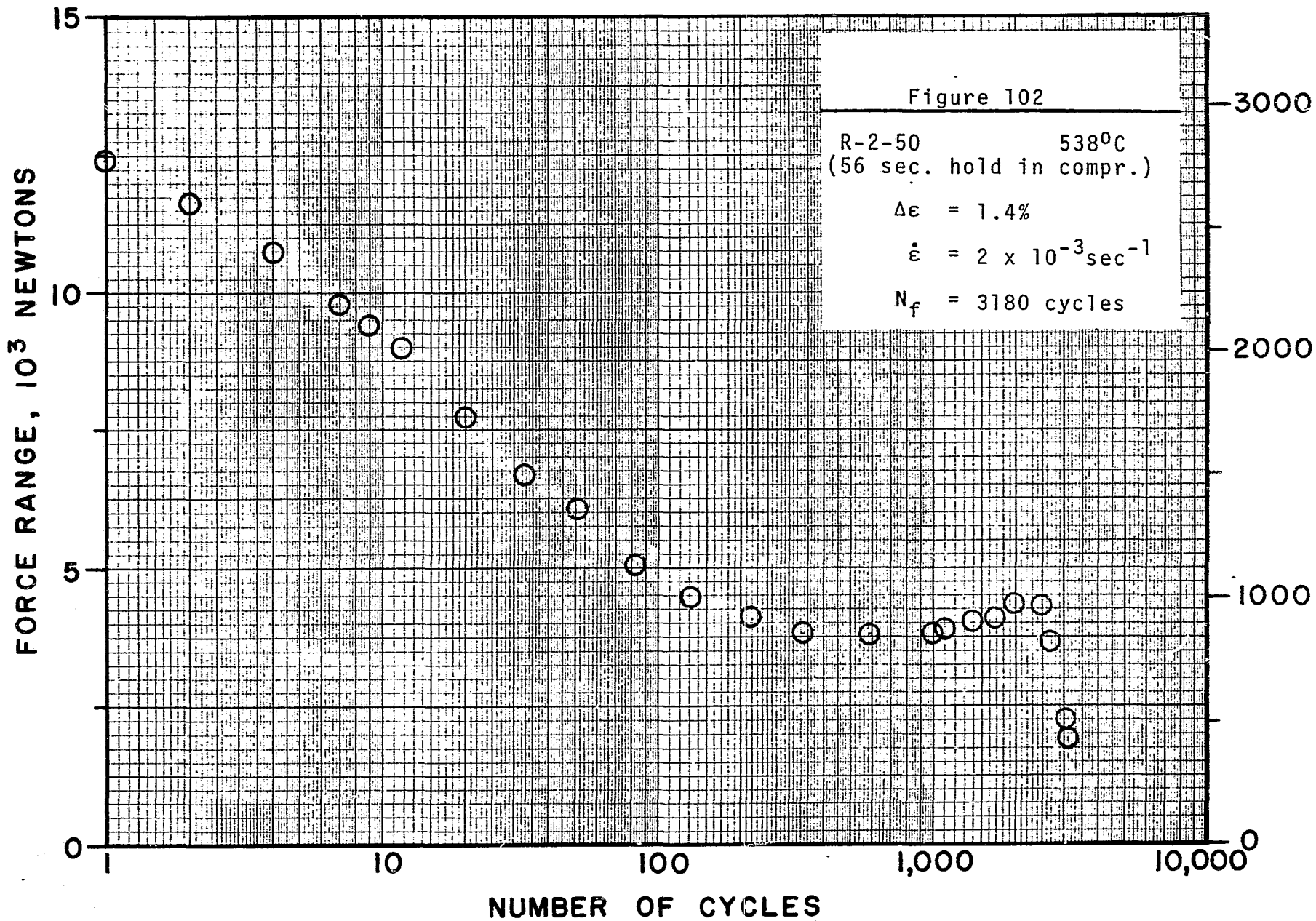


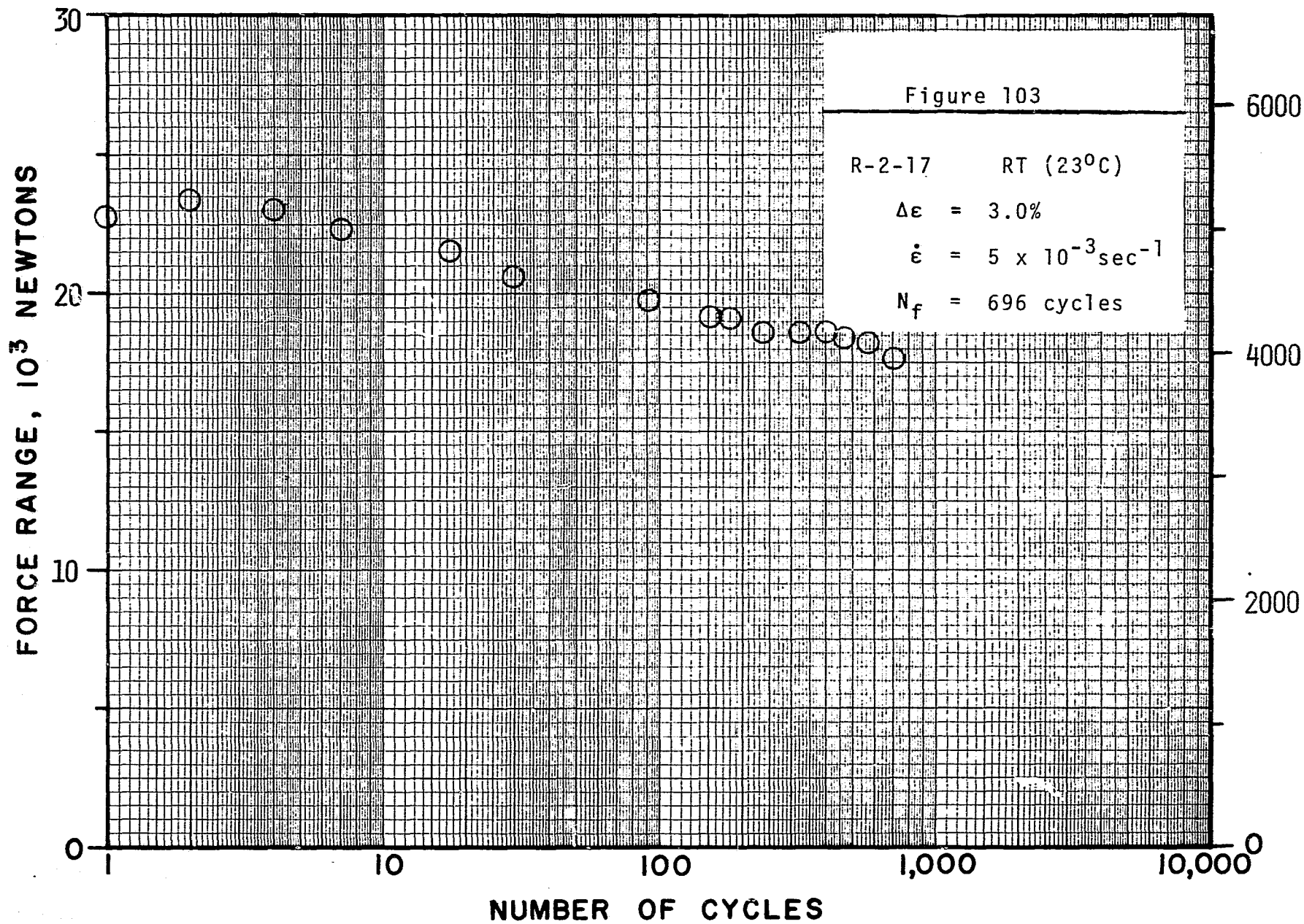


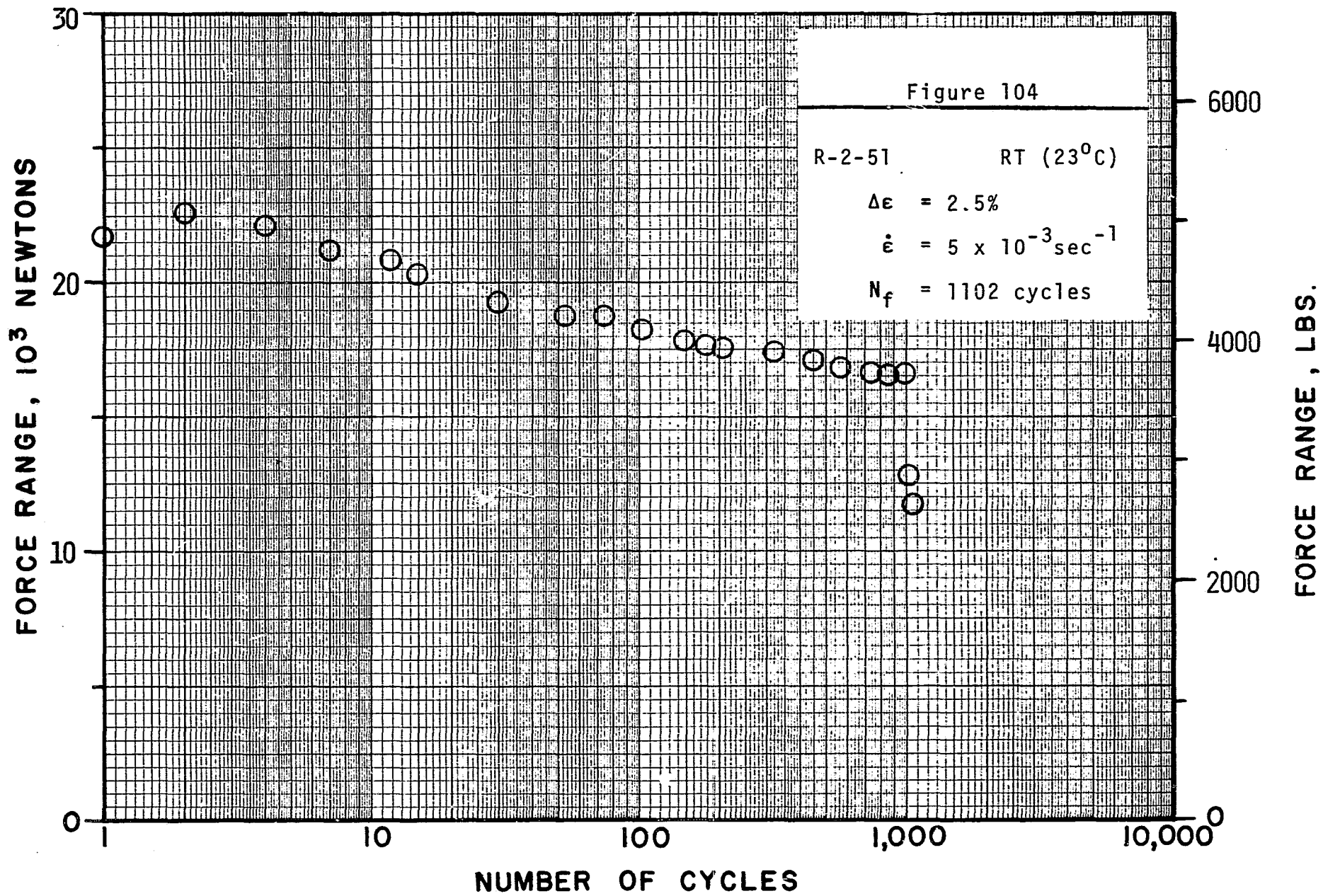


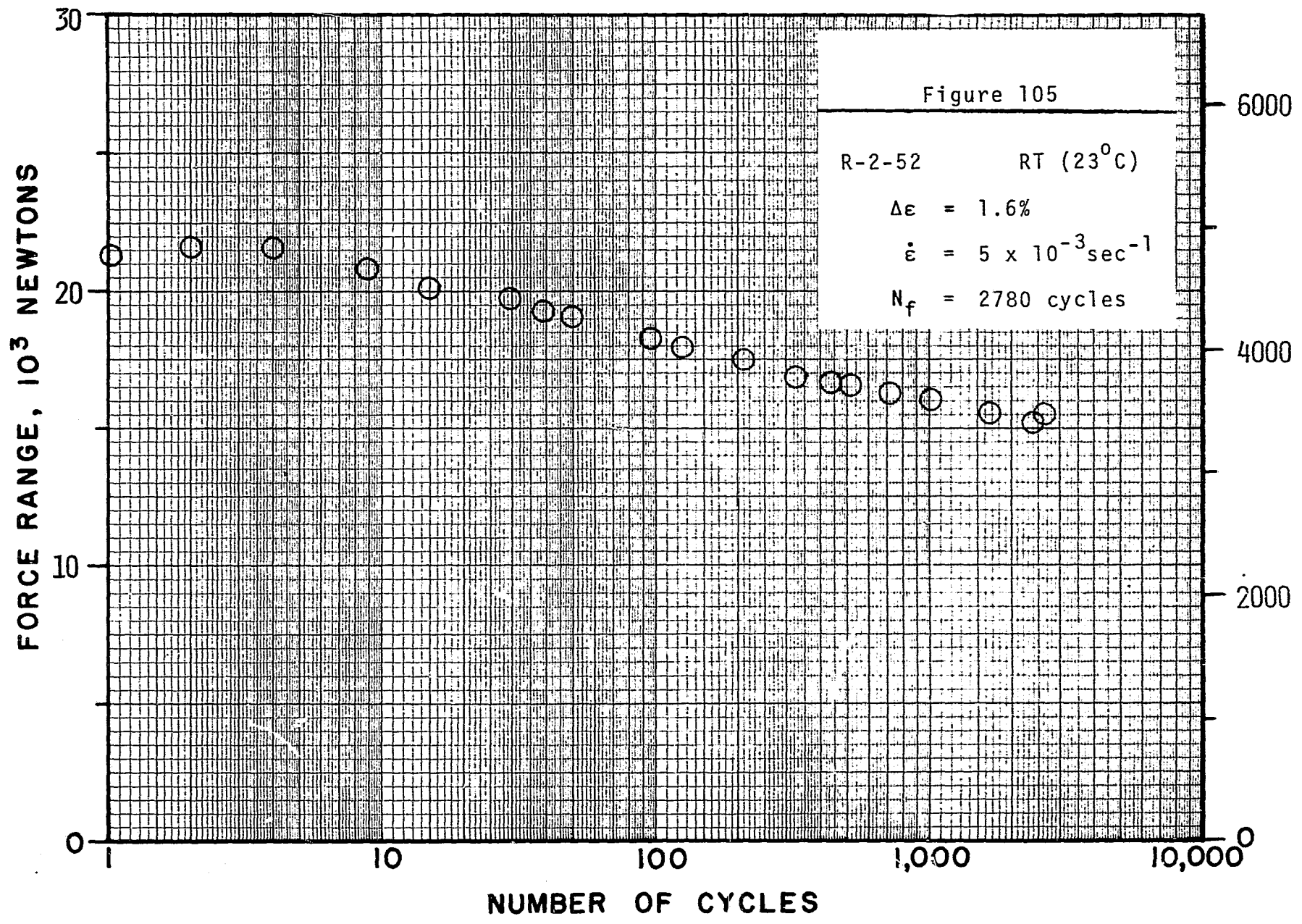
FORCE RANGE, LBS.

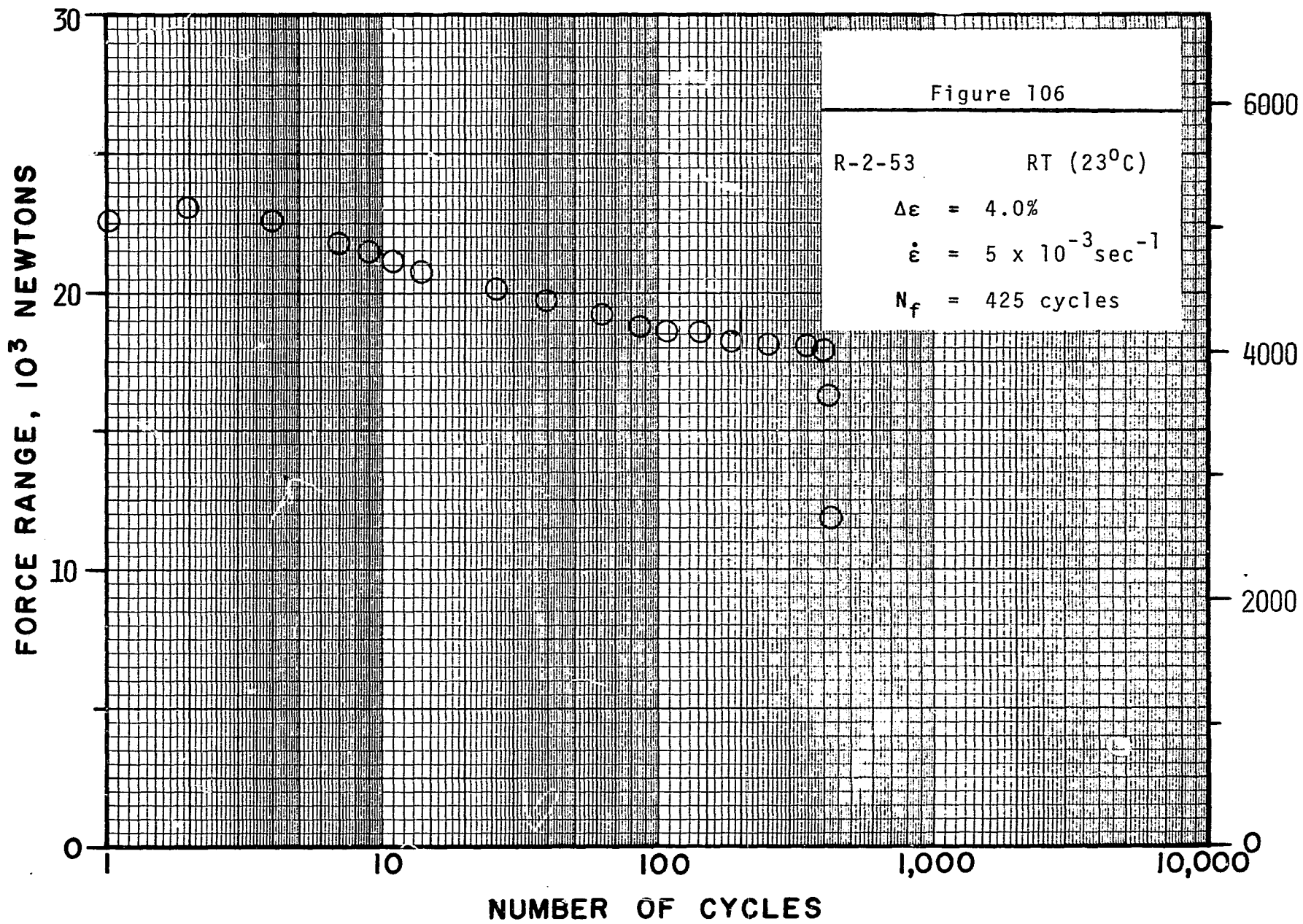


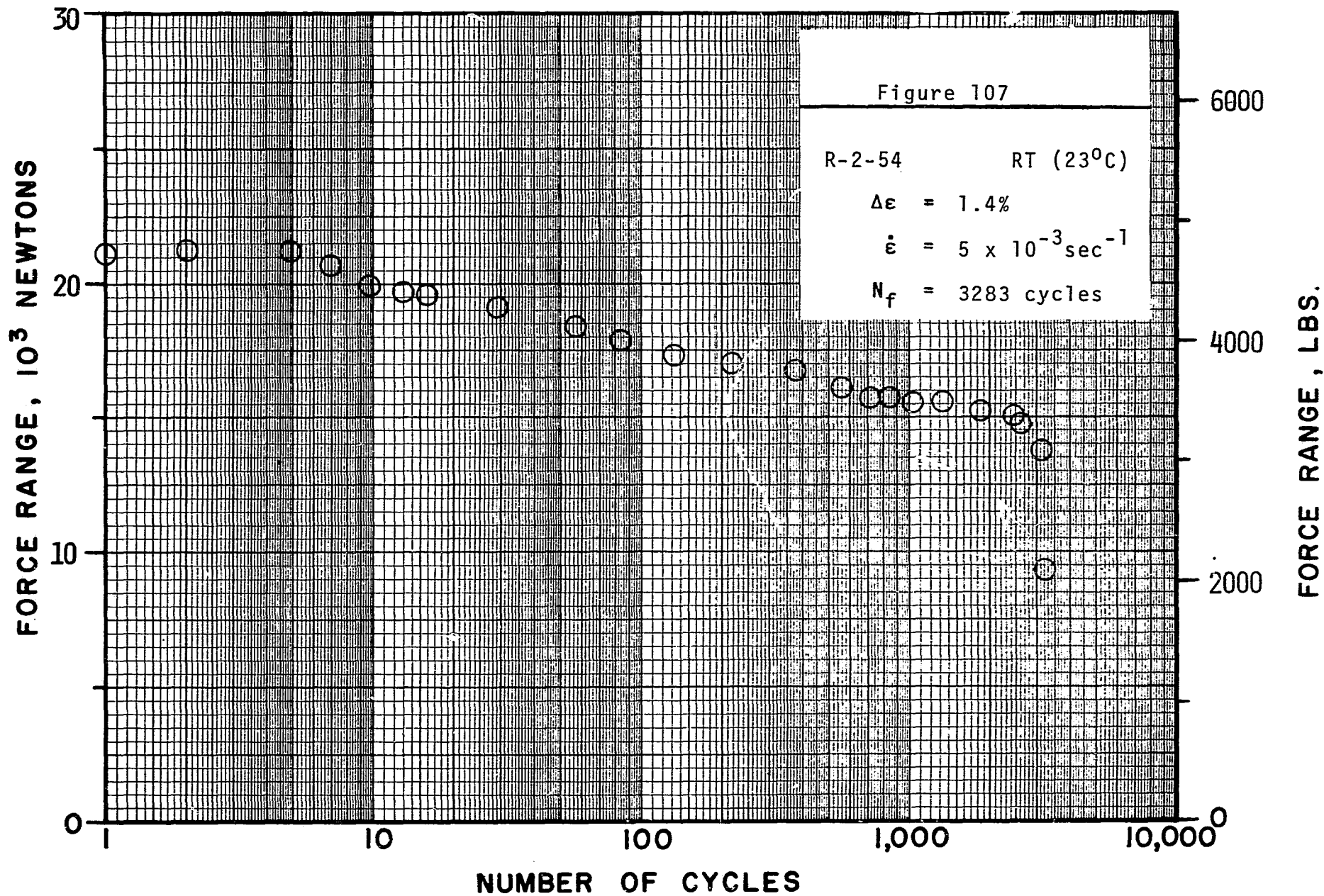


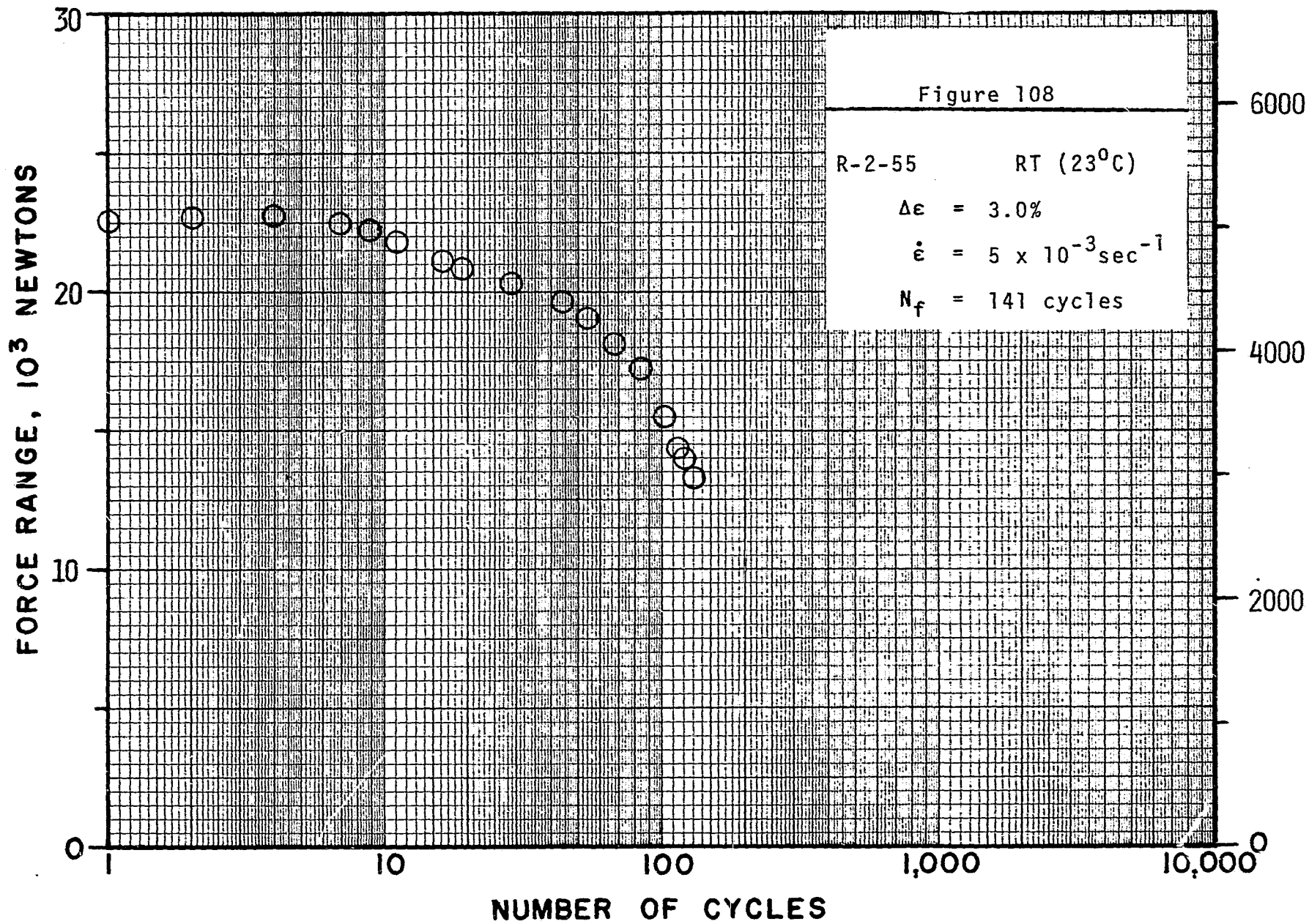




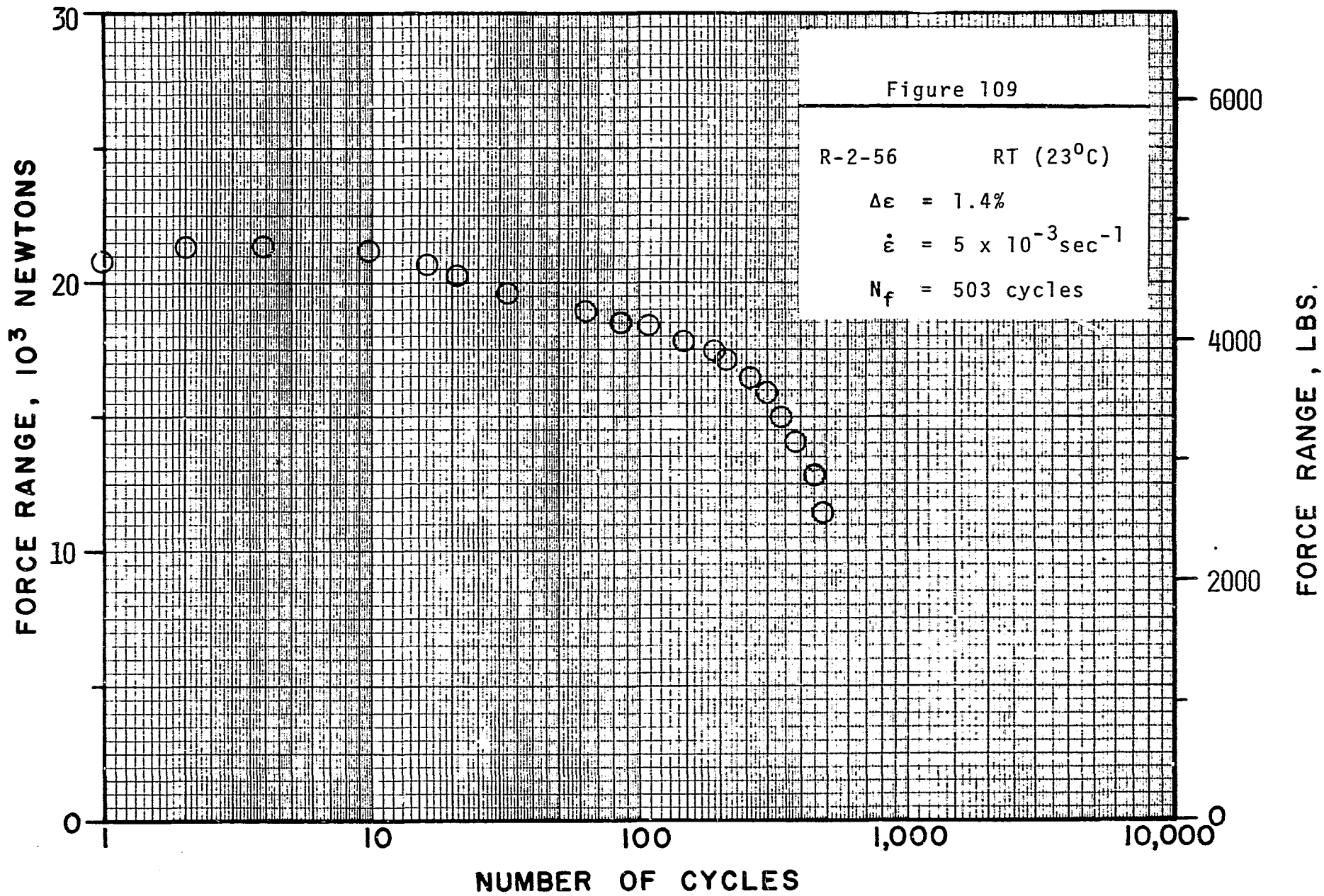


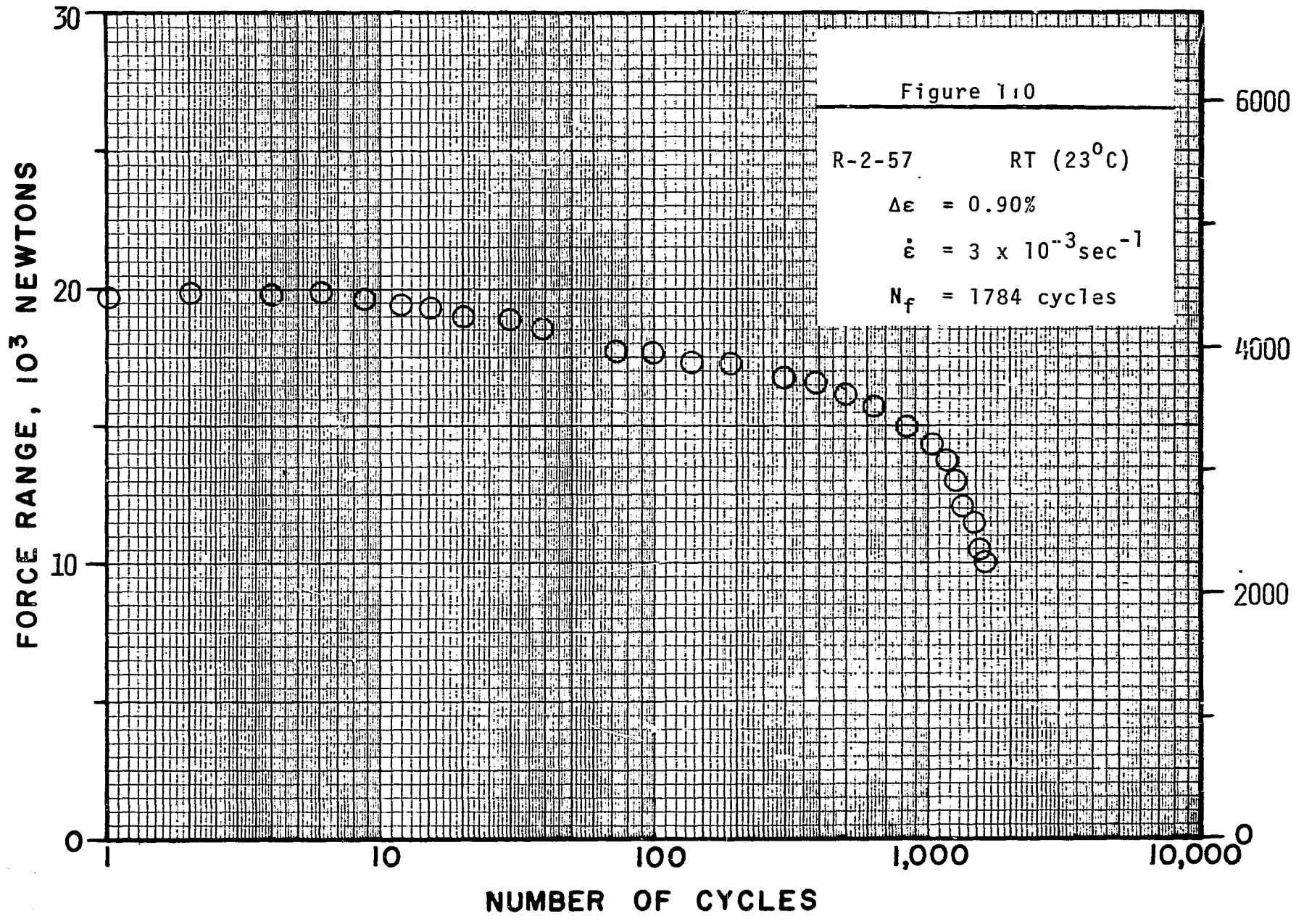


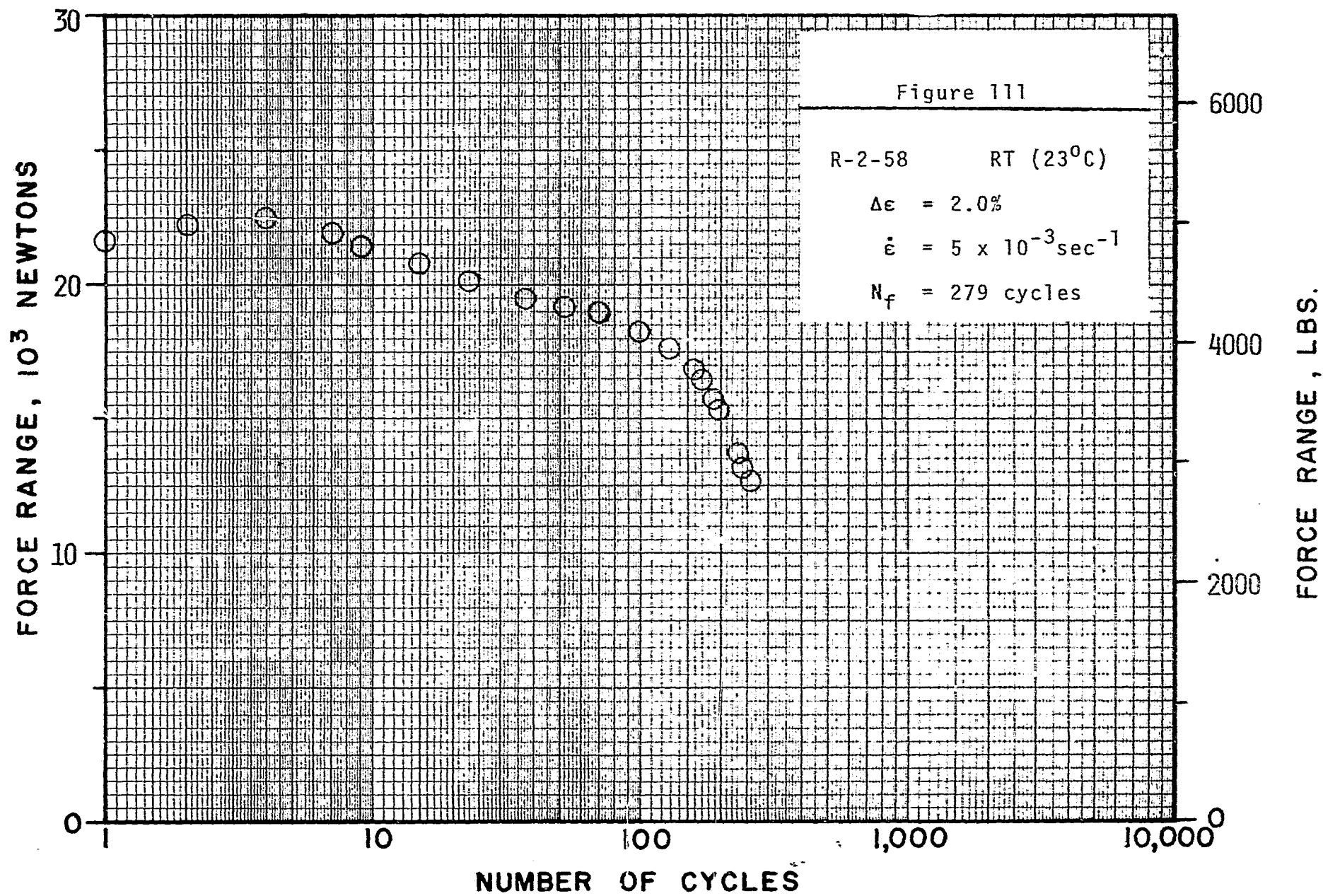


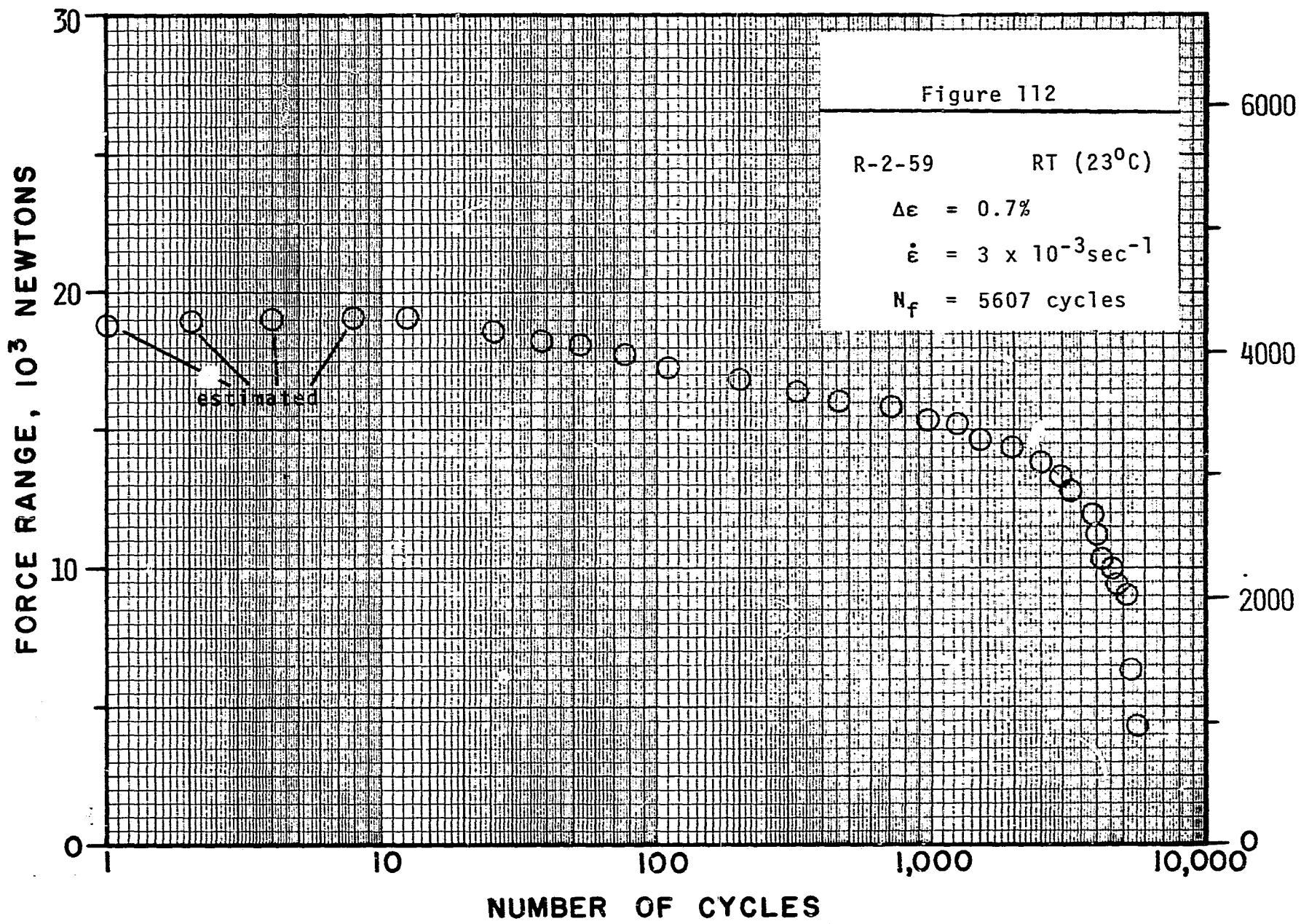


FORCE RANGE, LBS.

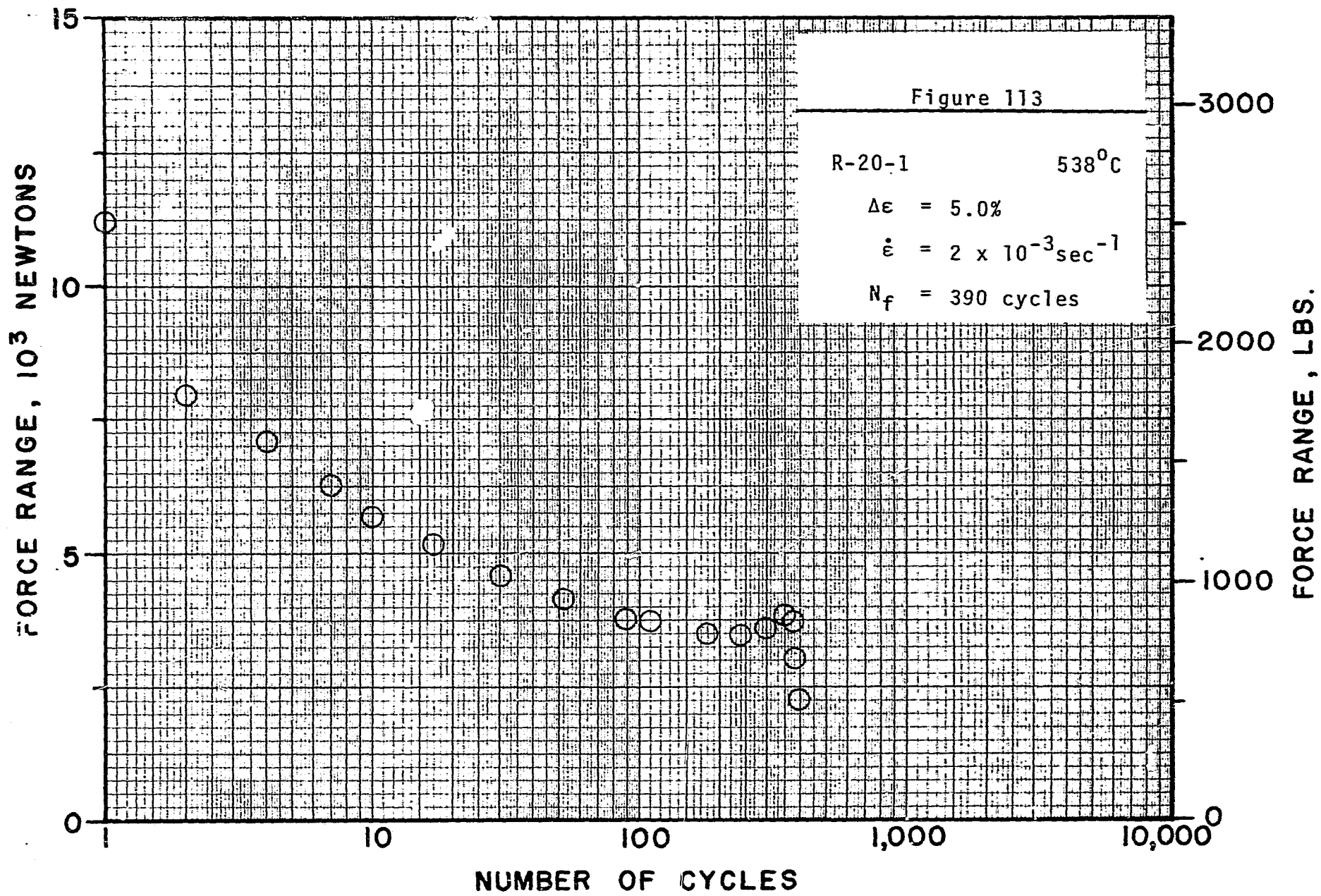


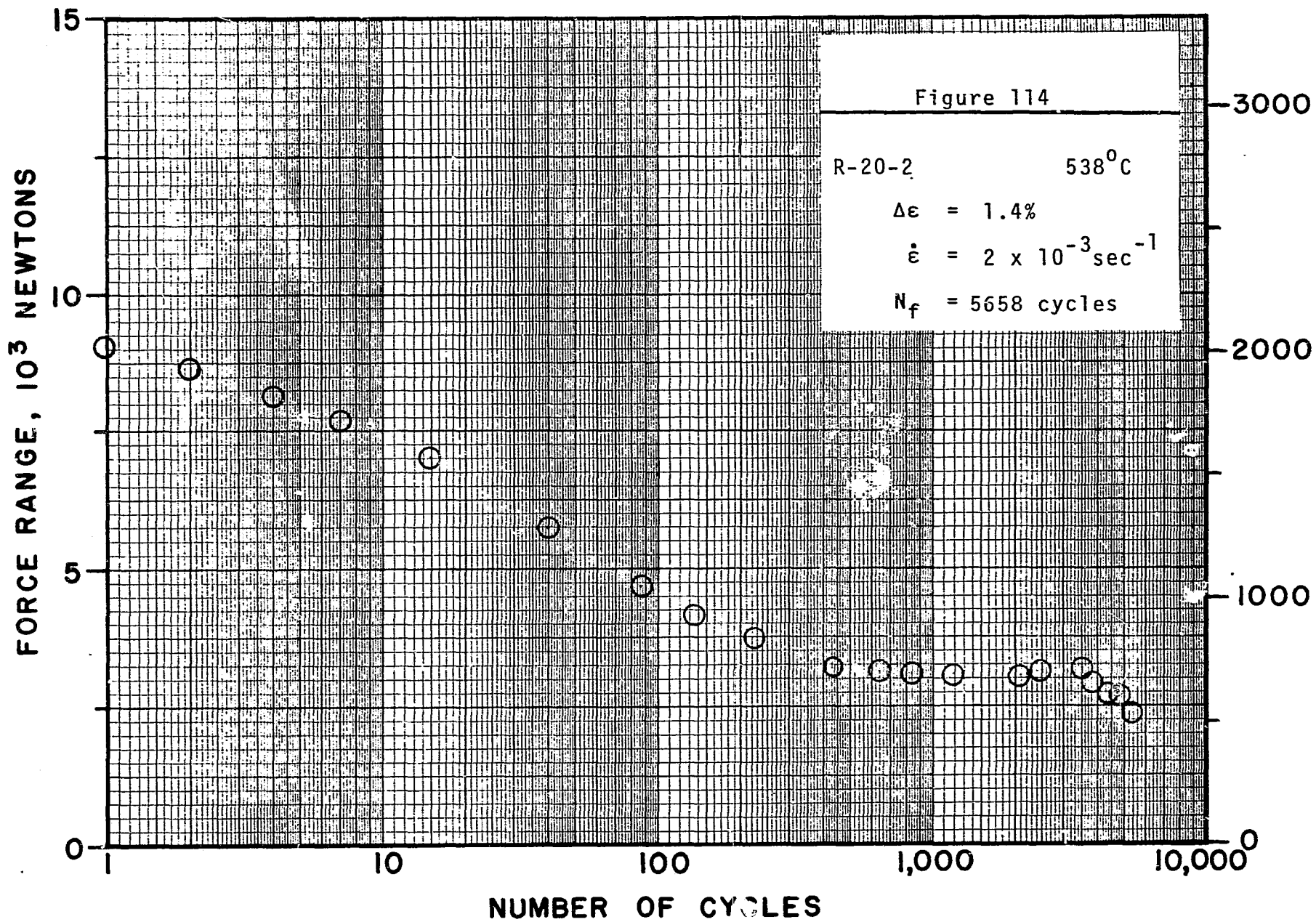


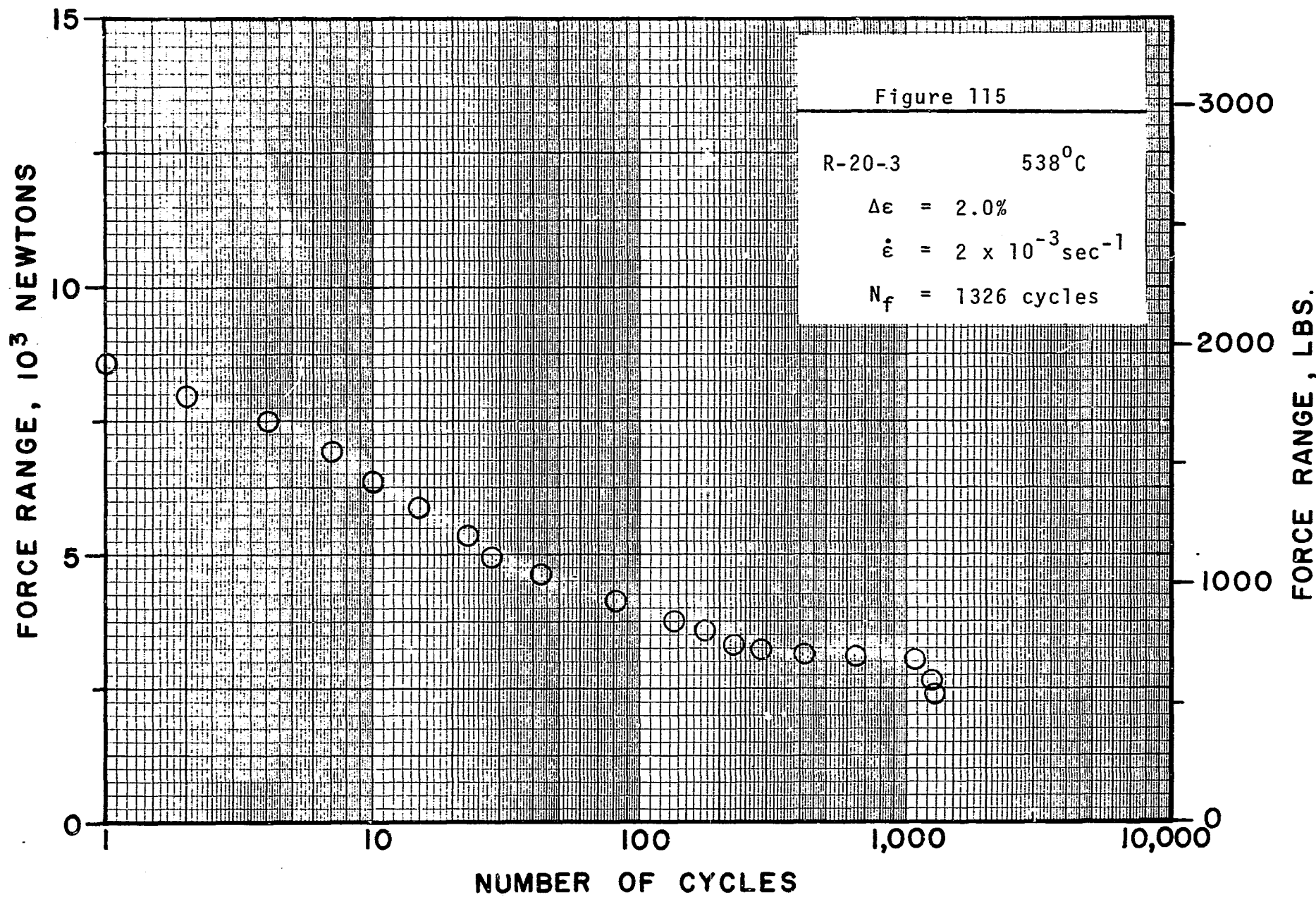


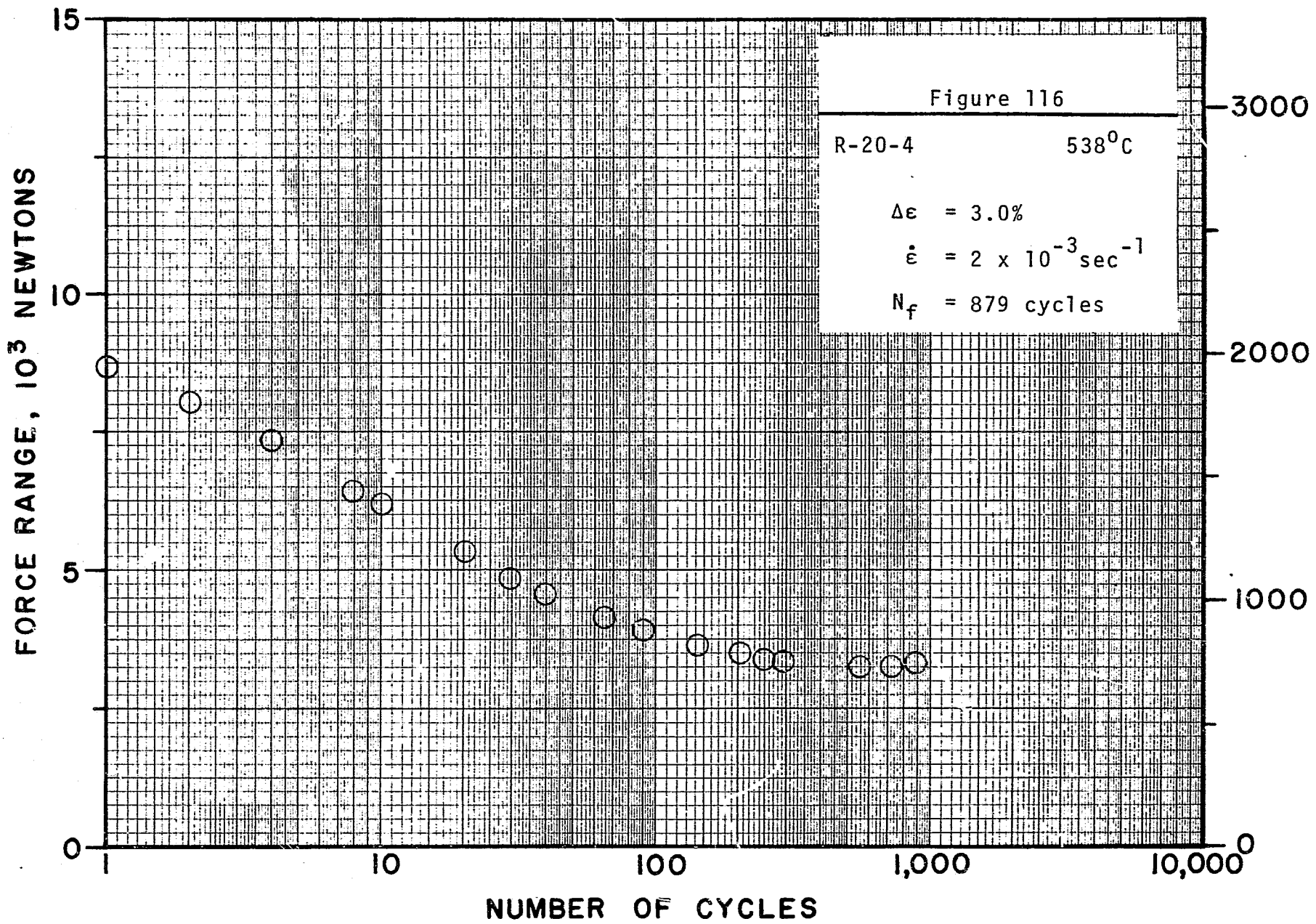


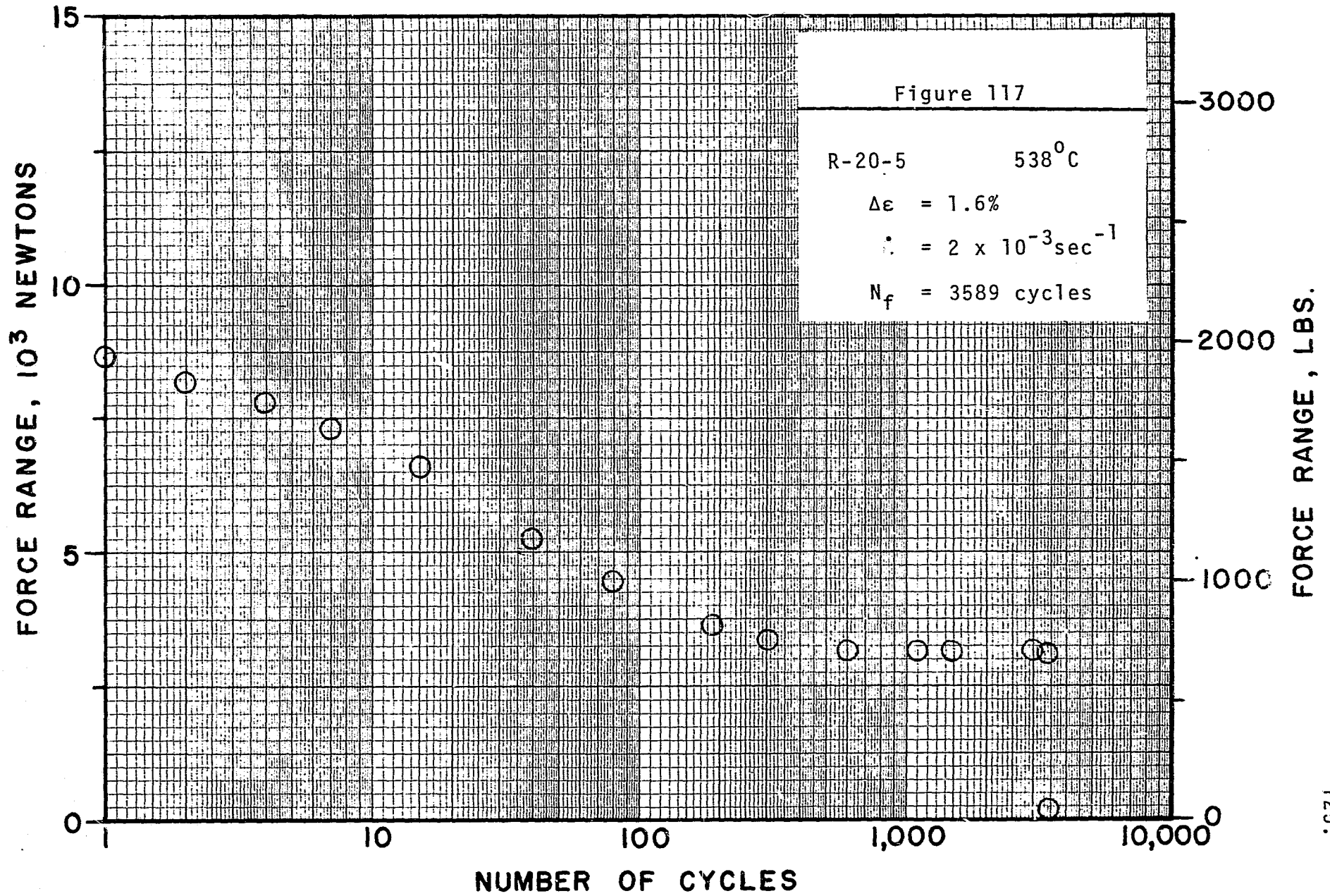
FORCE RANGE, LBS.

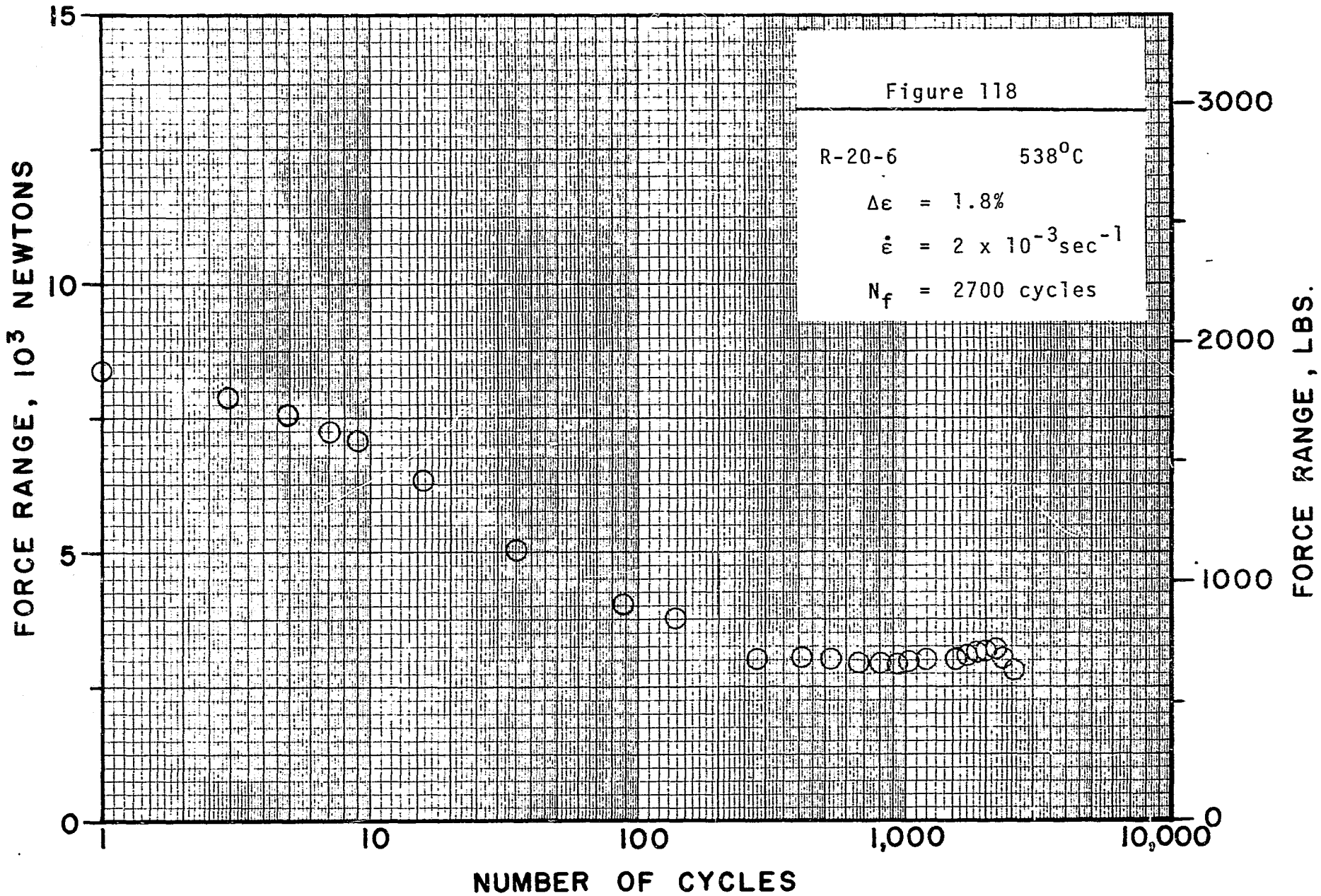


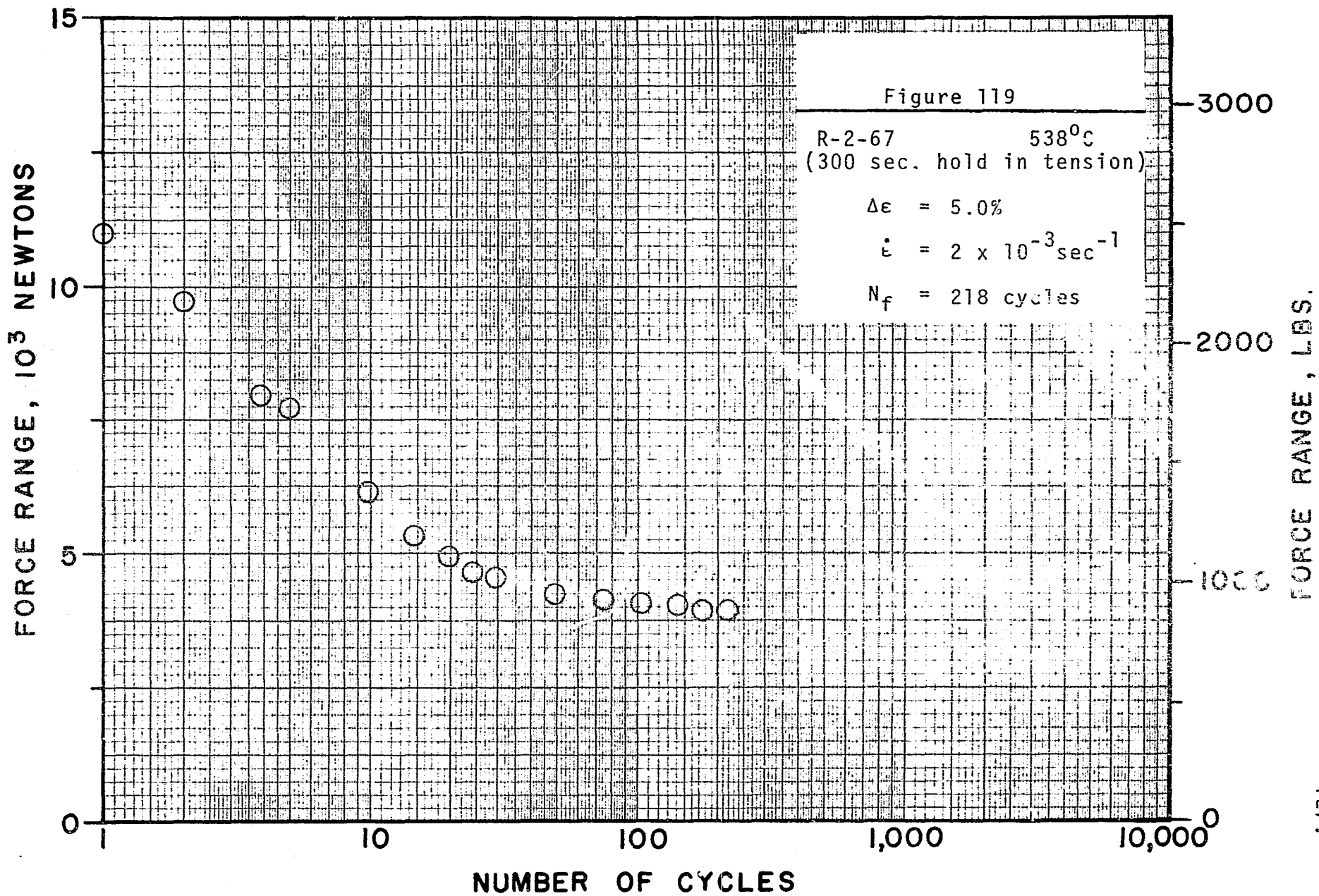


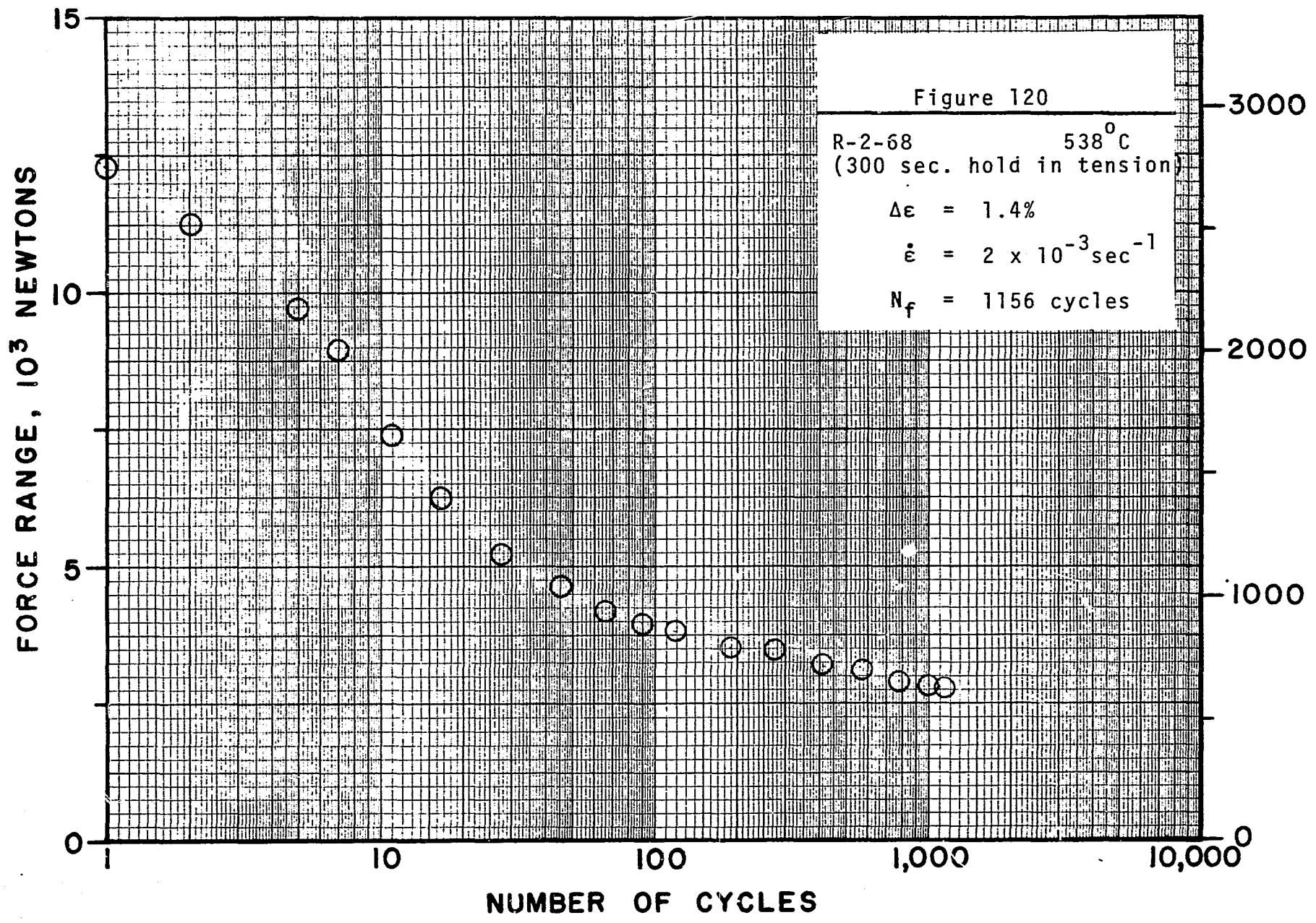




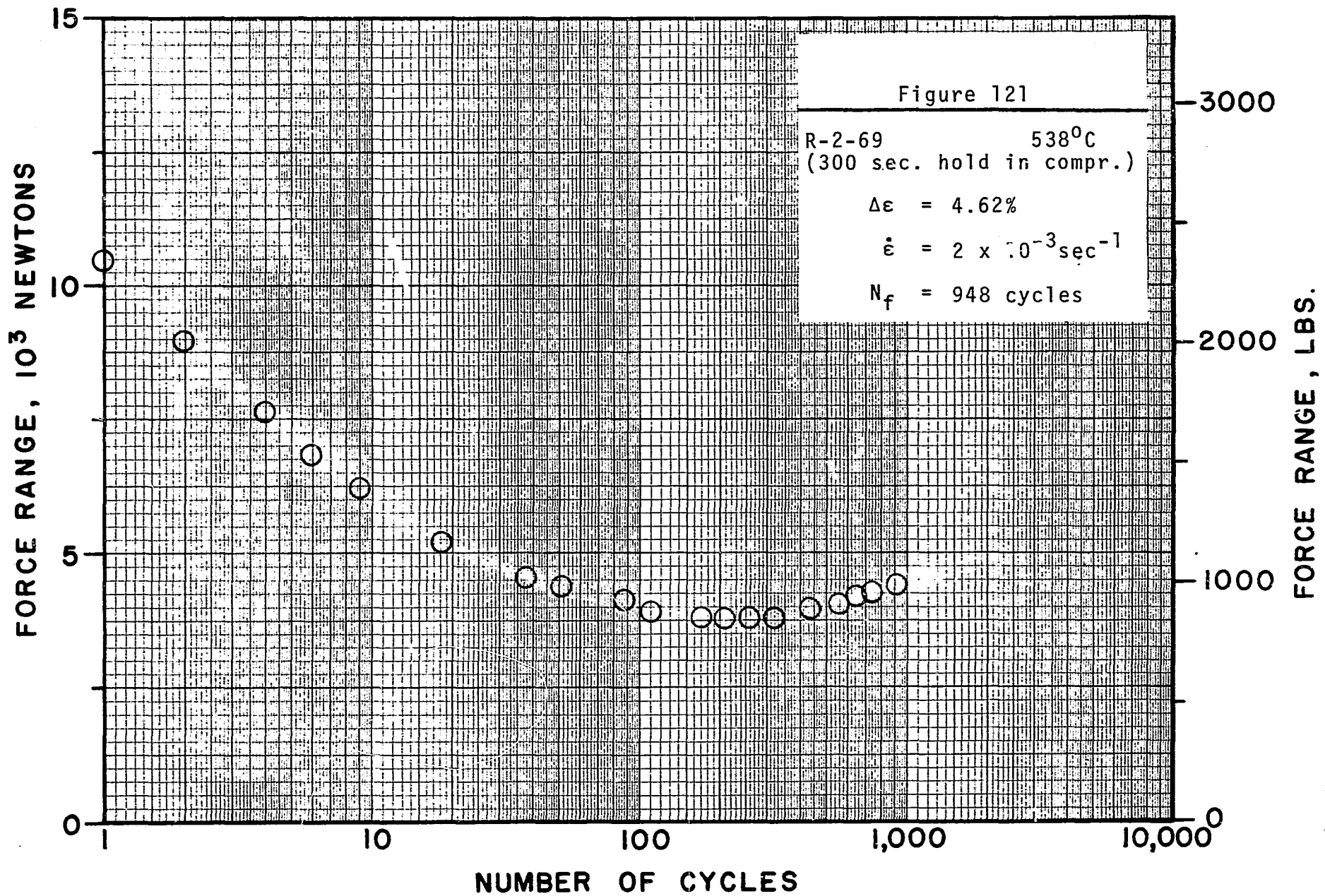


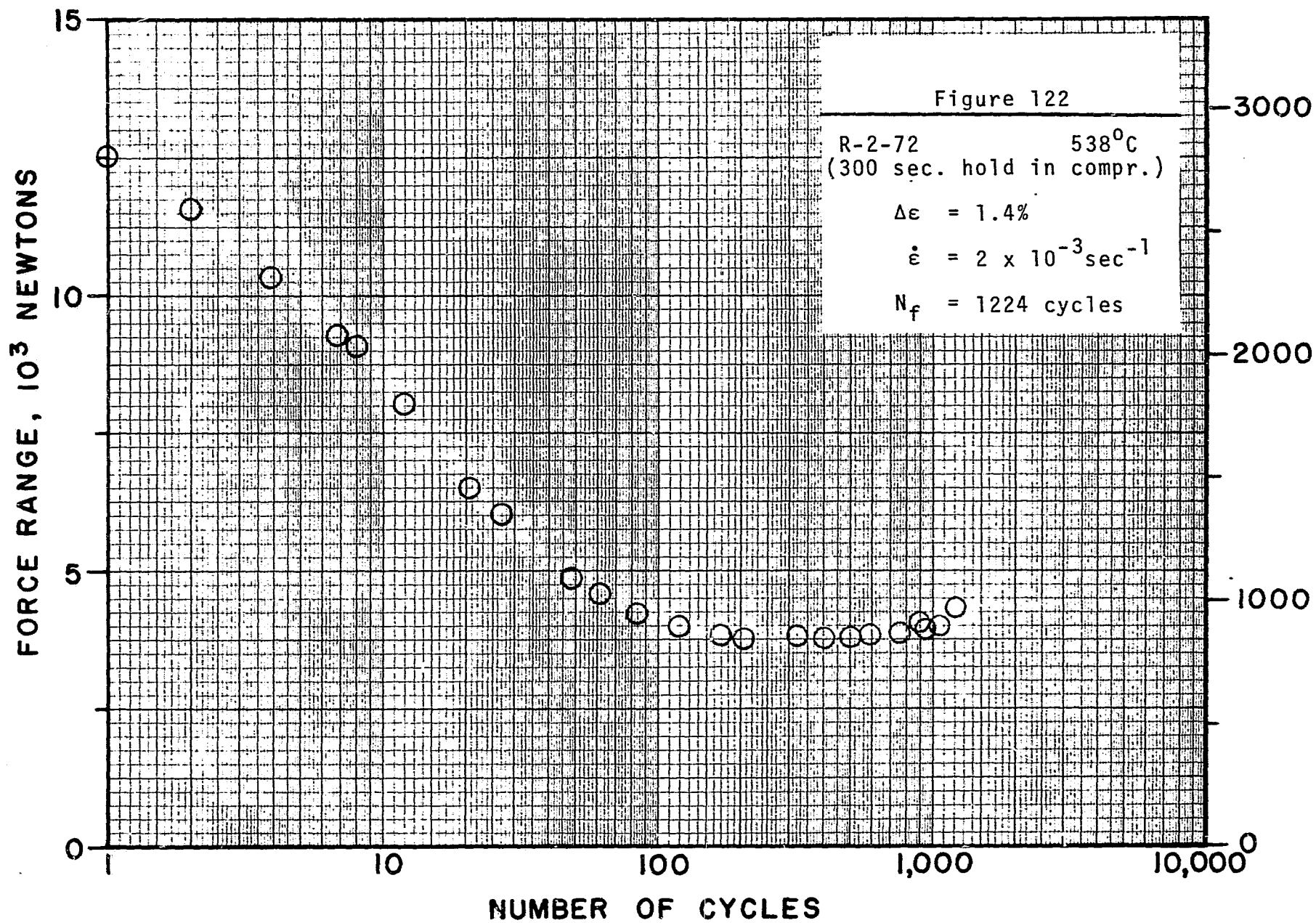




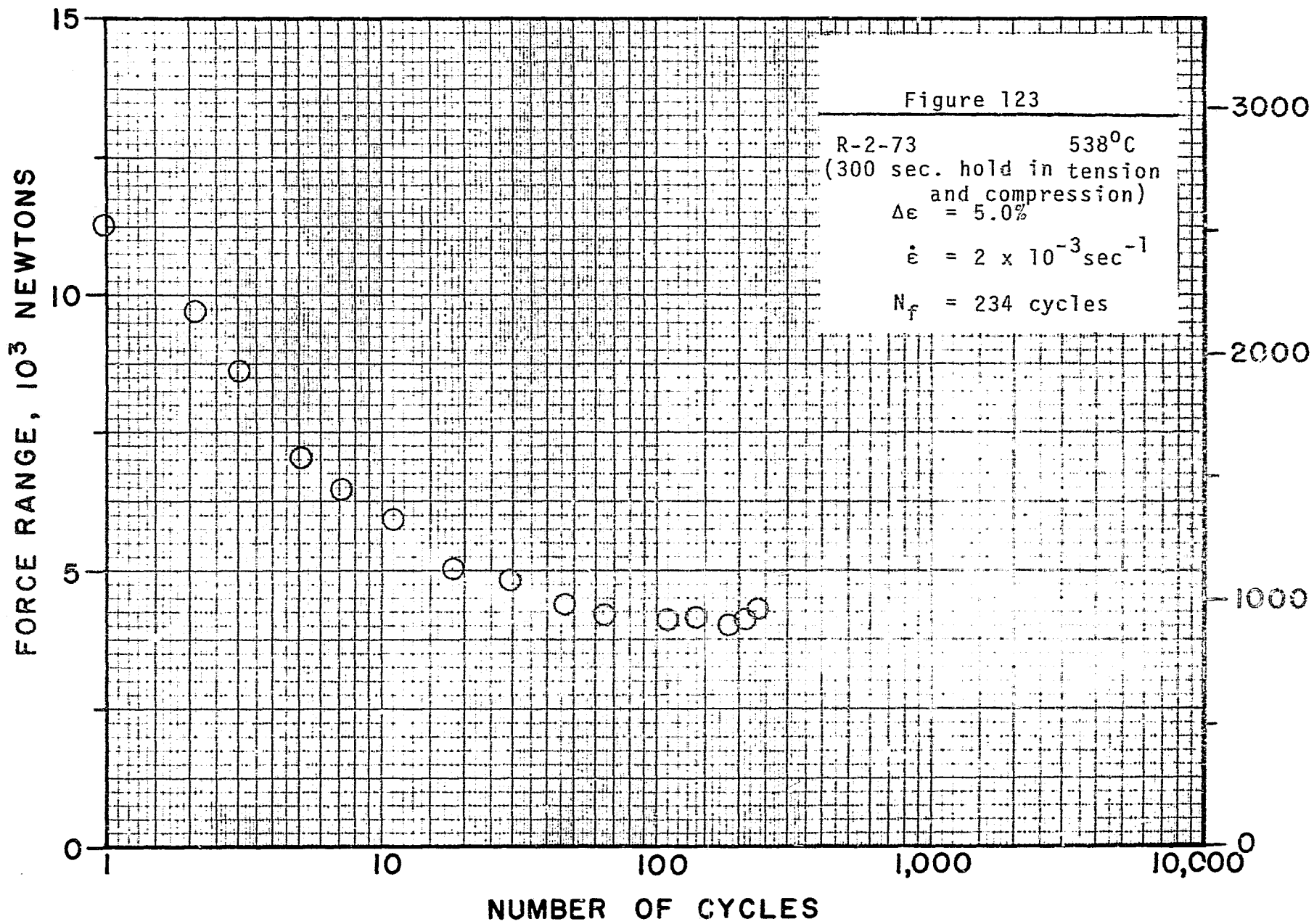


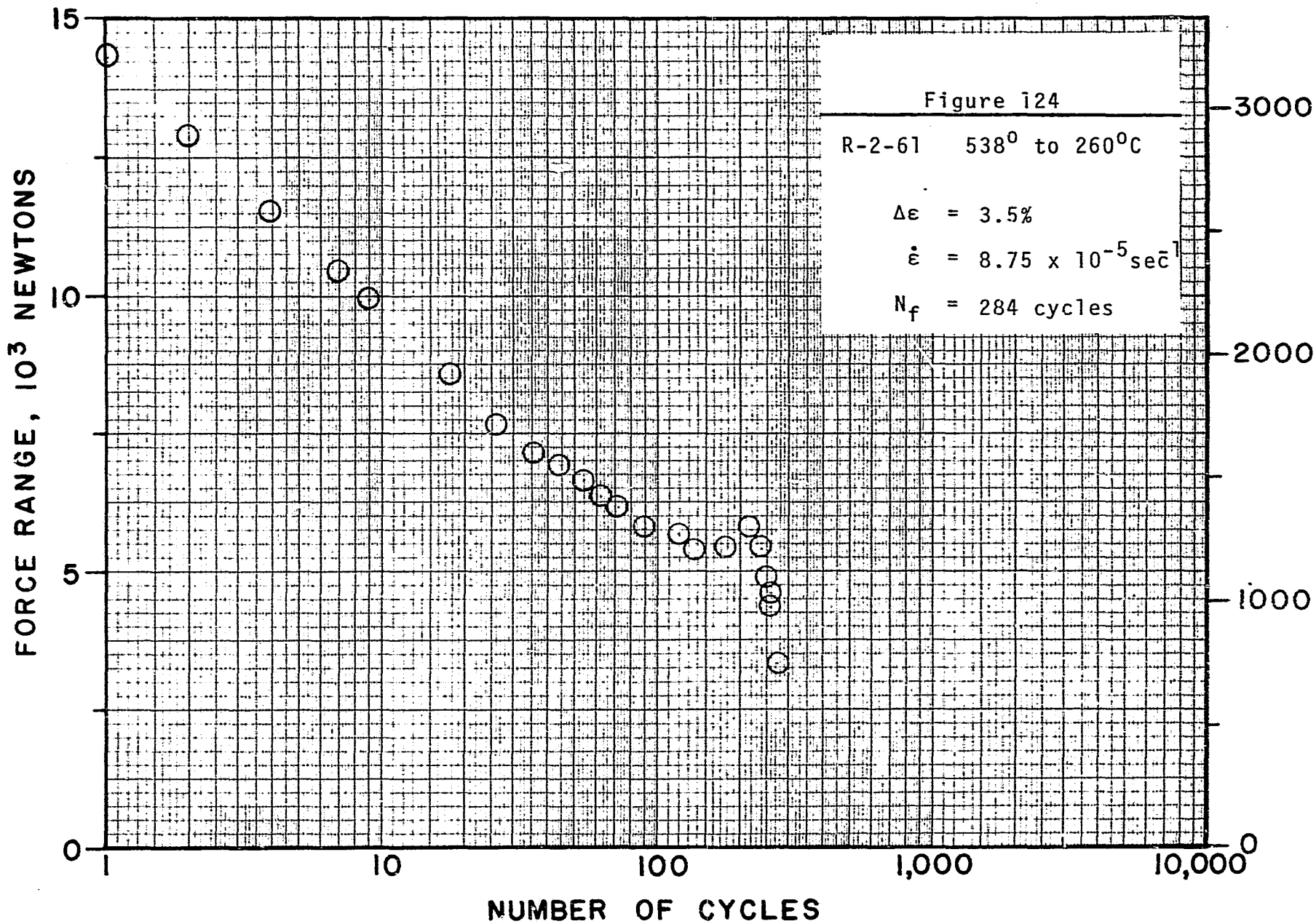
FORCE RANGE, LBS.

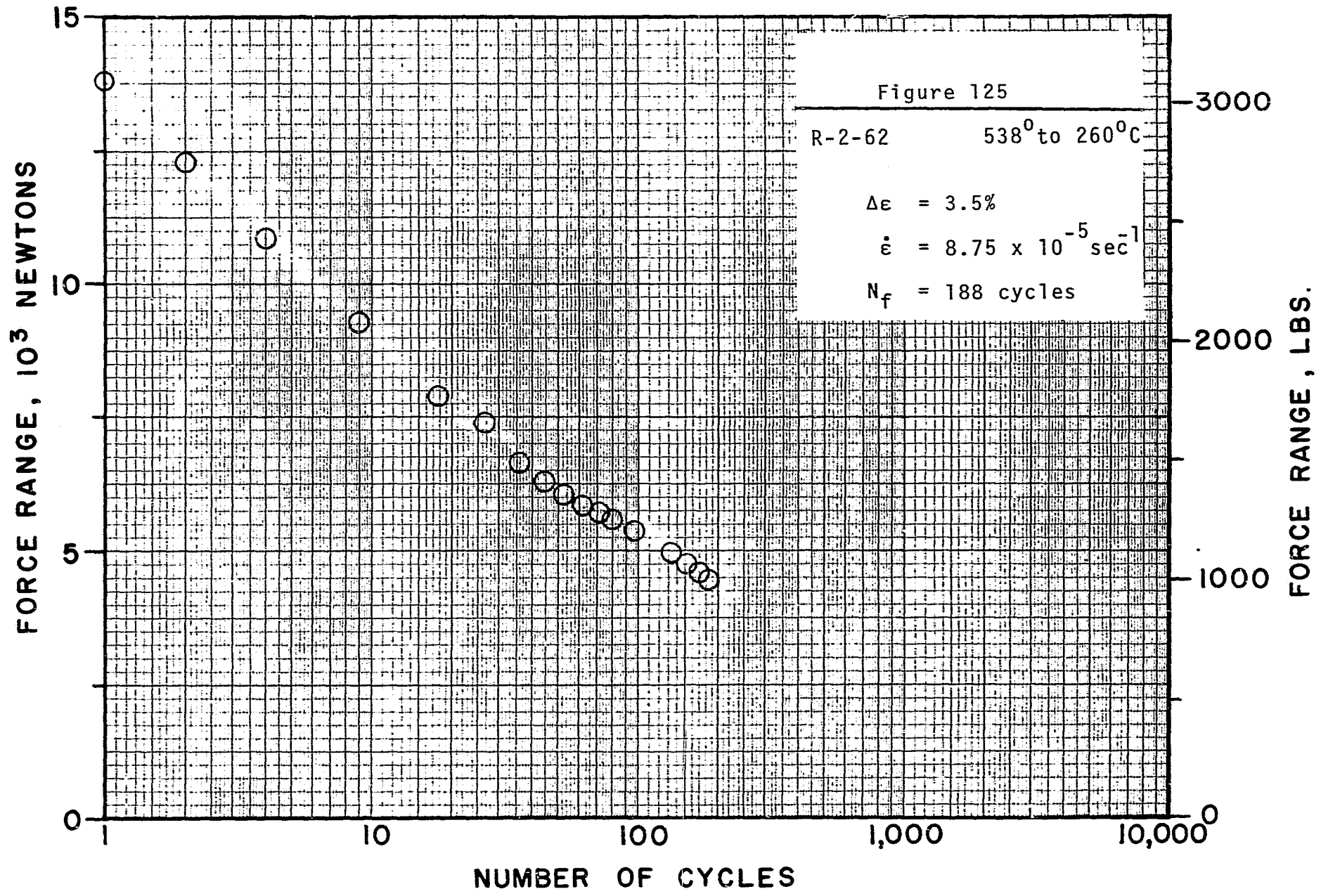


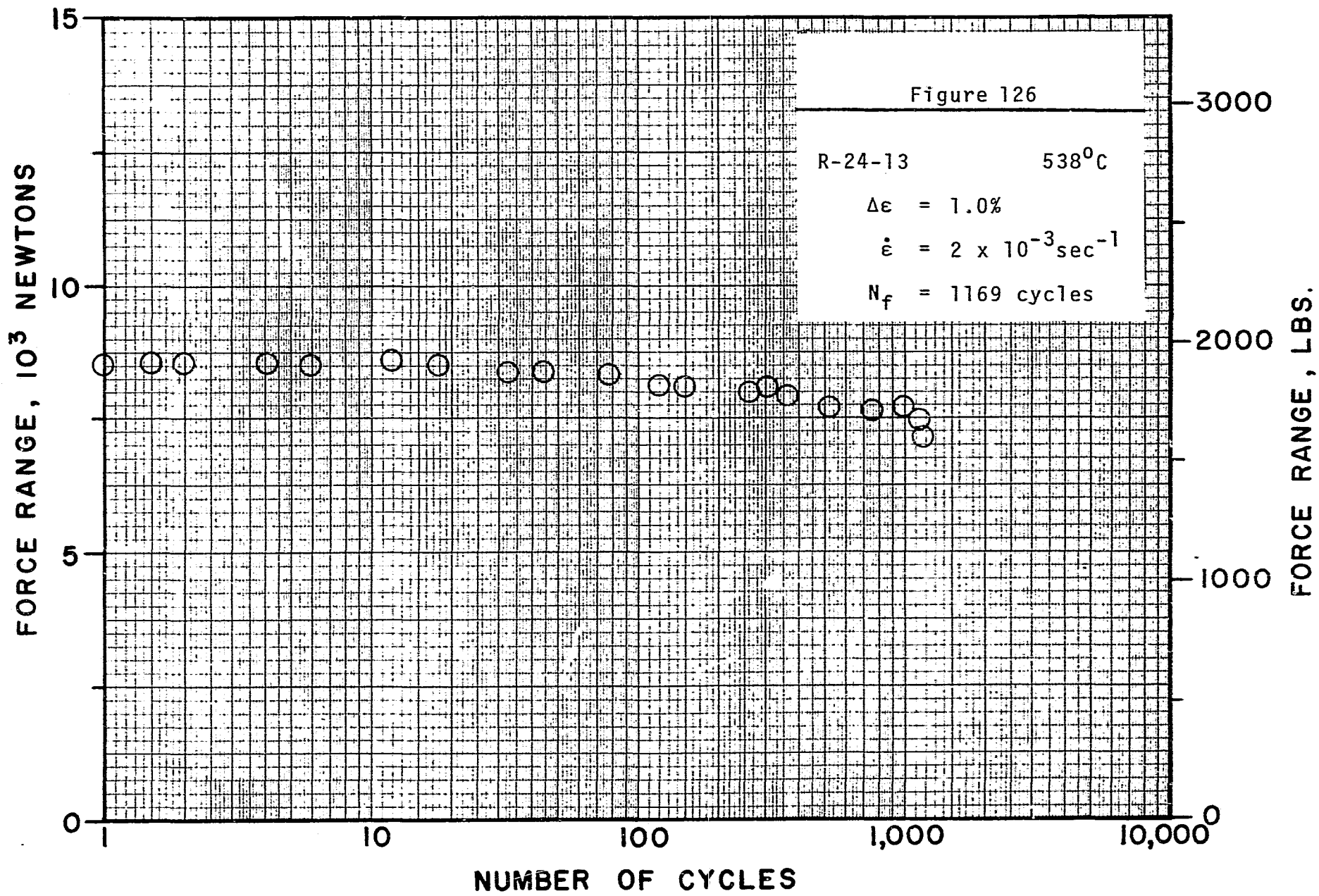


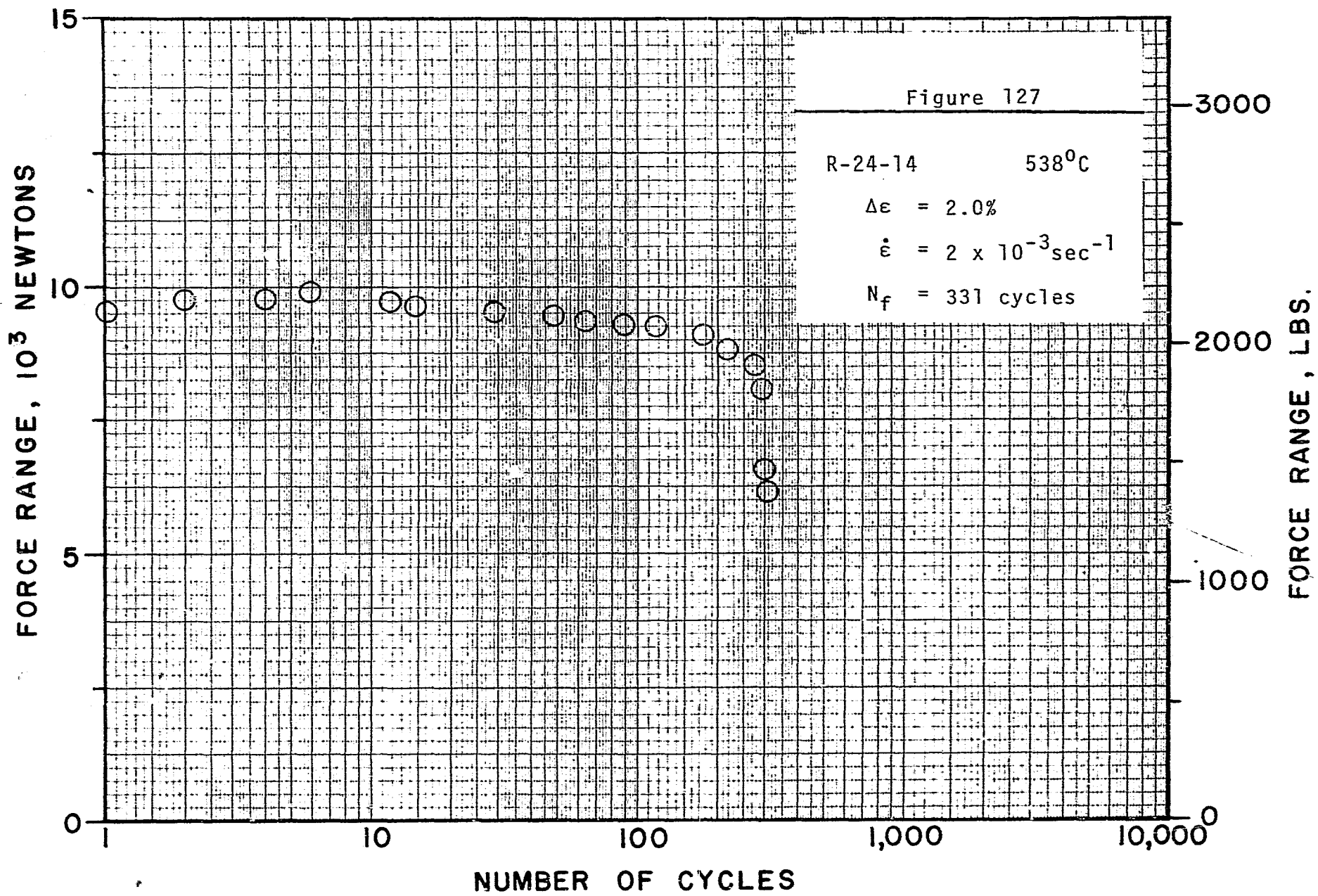
FORCE RANGE, LBS.

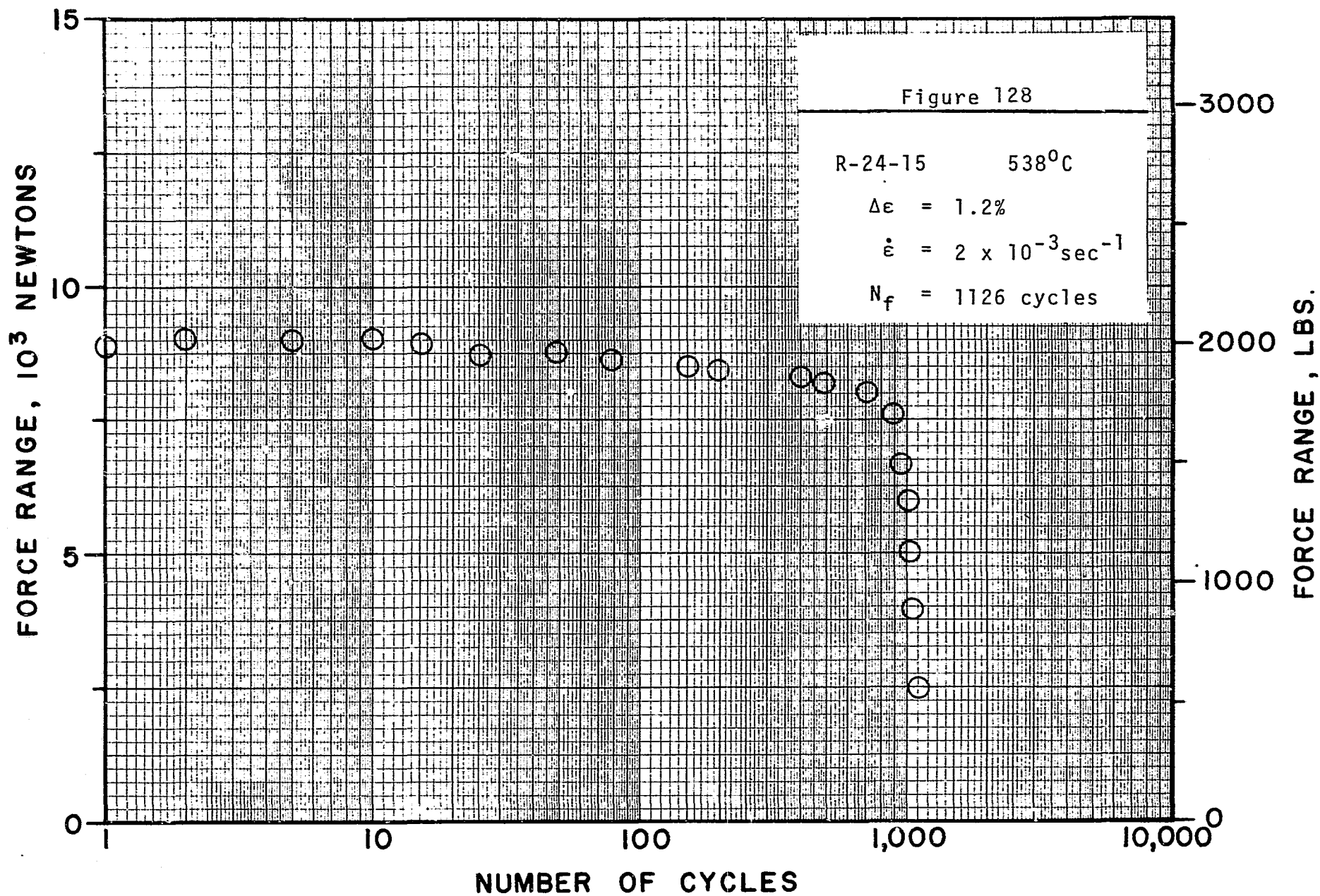


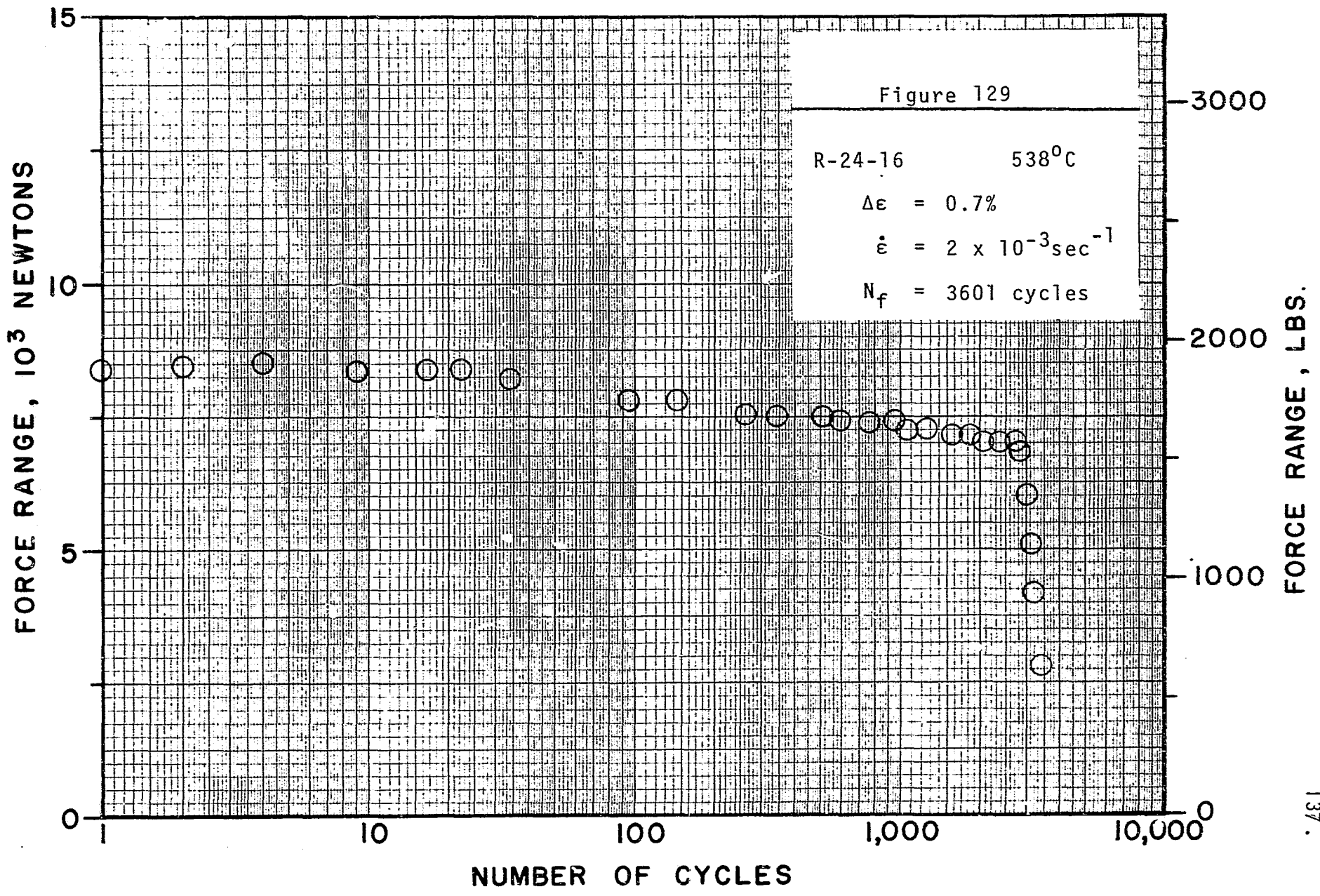


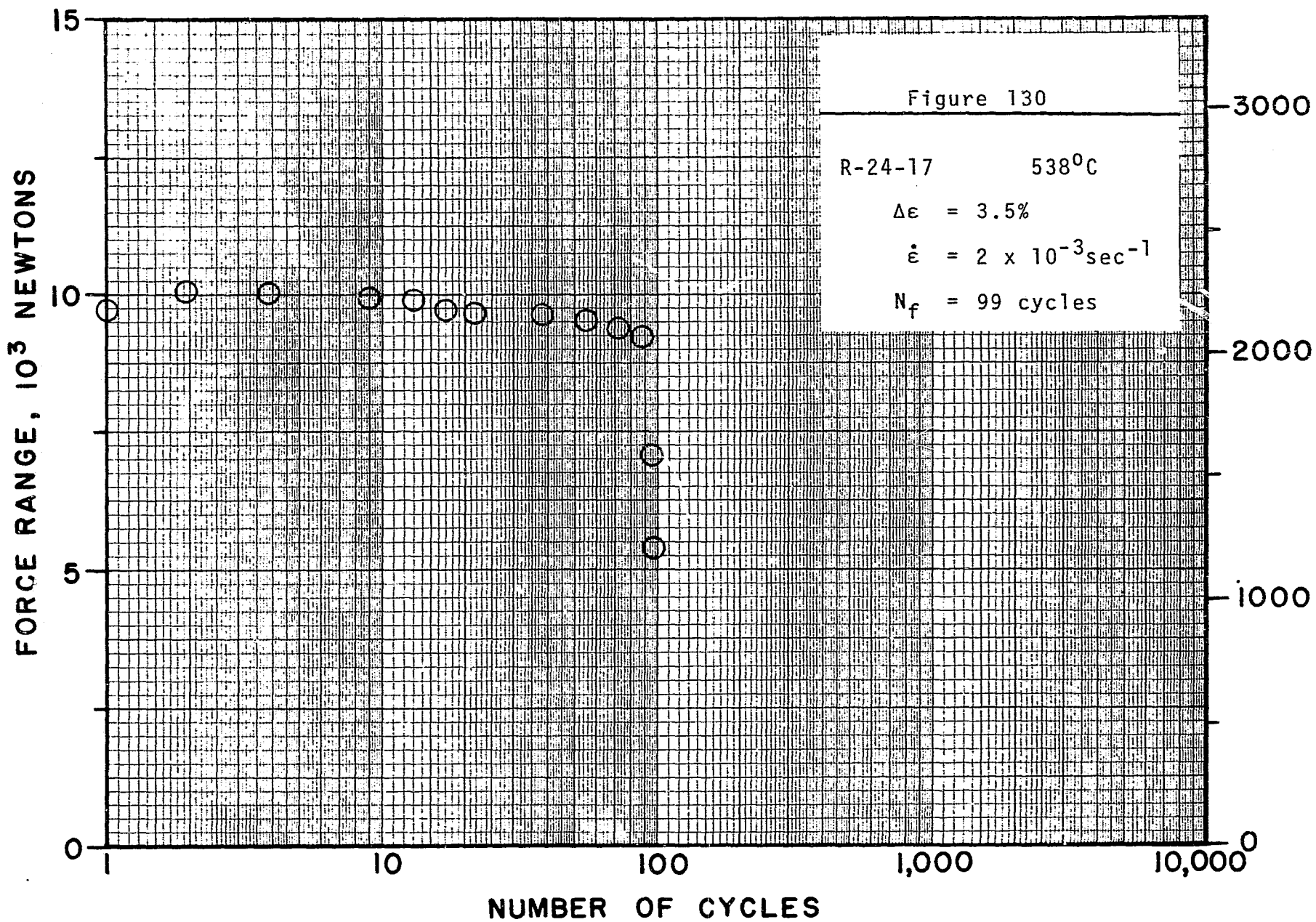


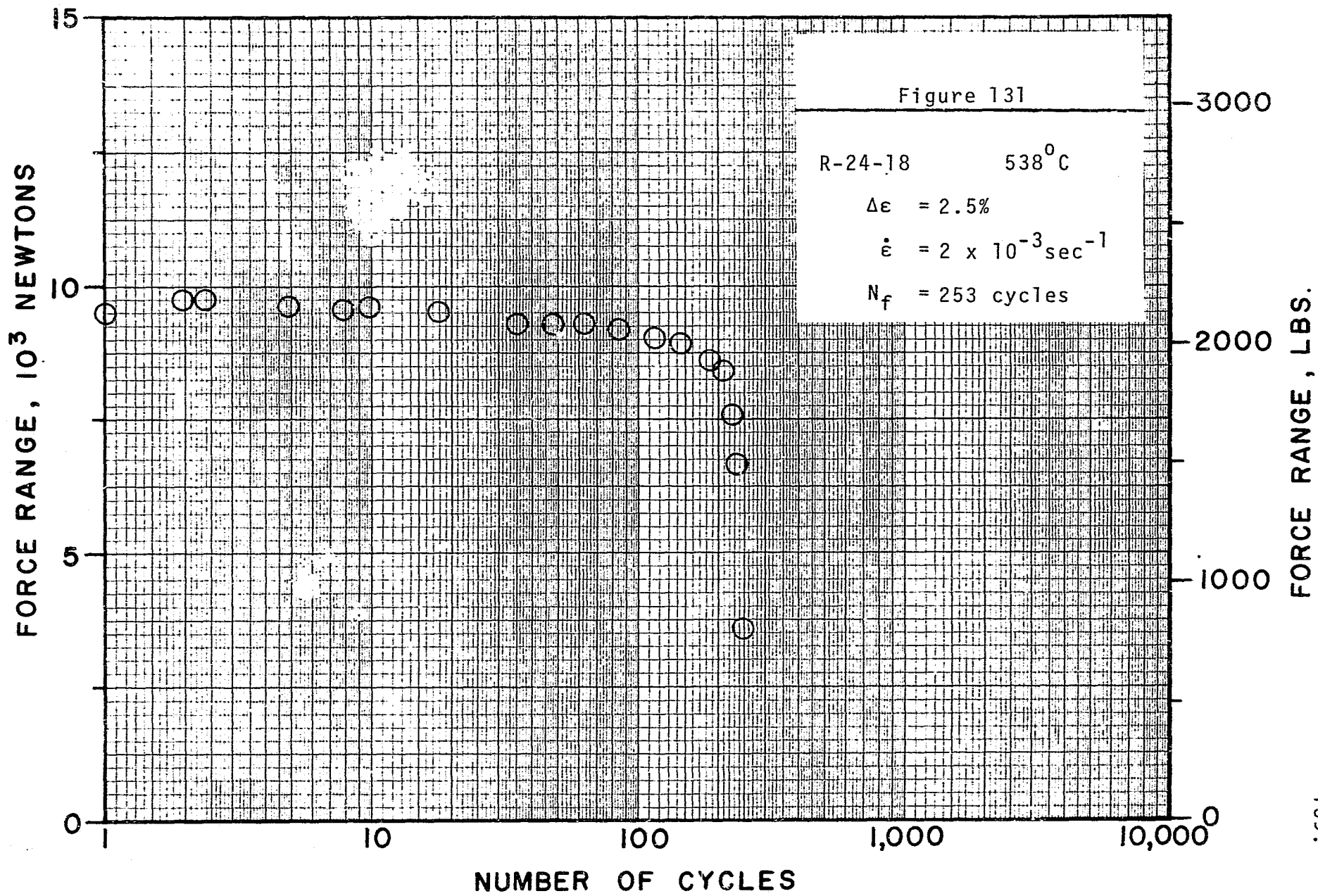


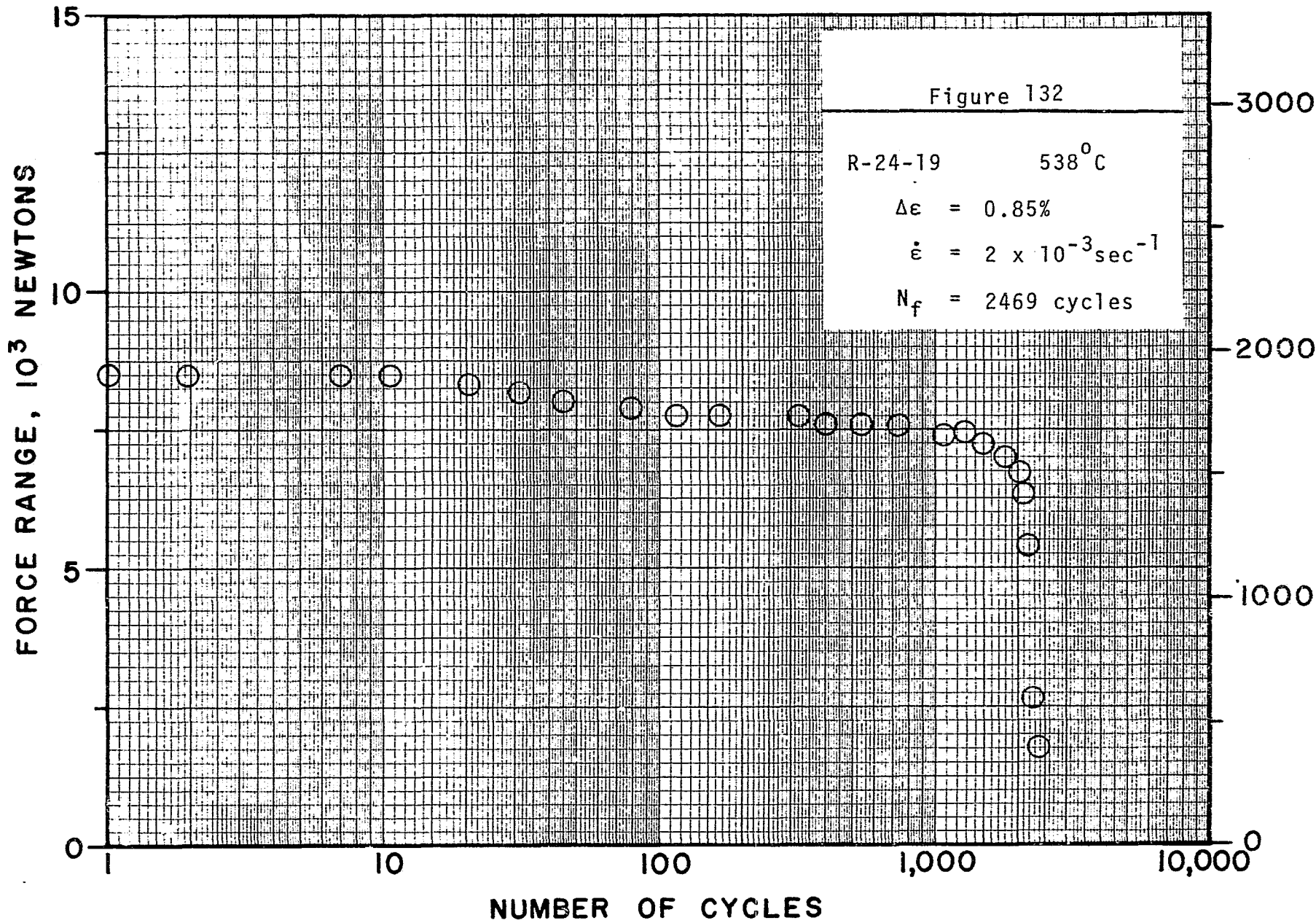


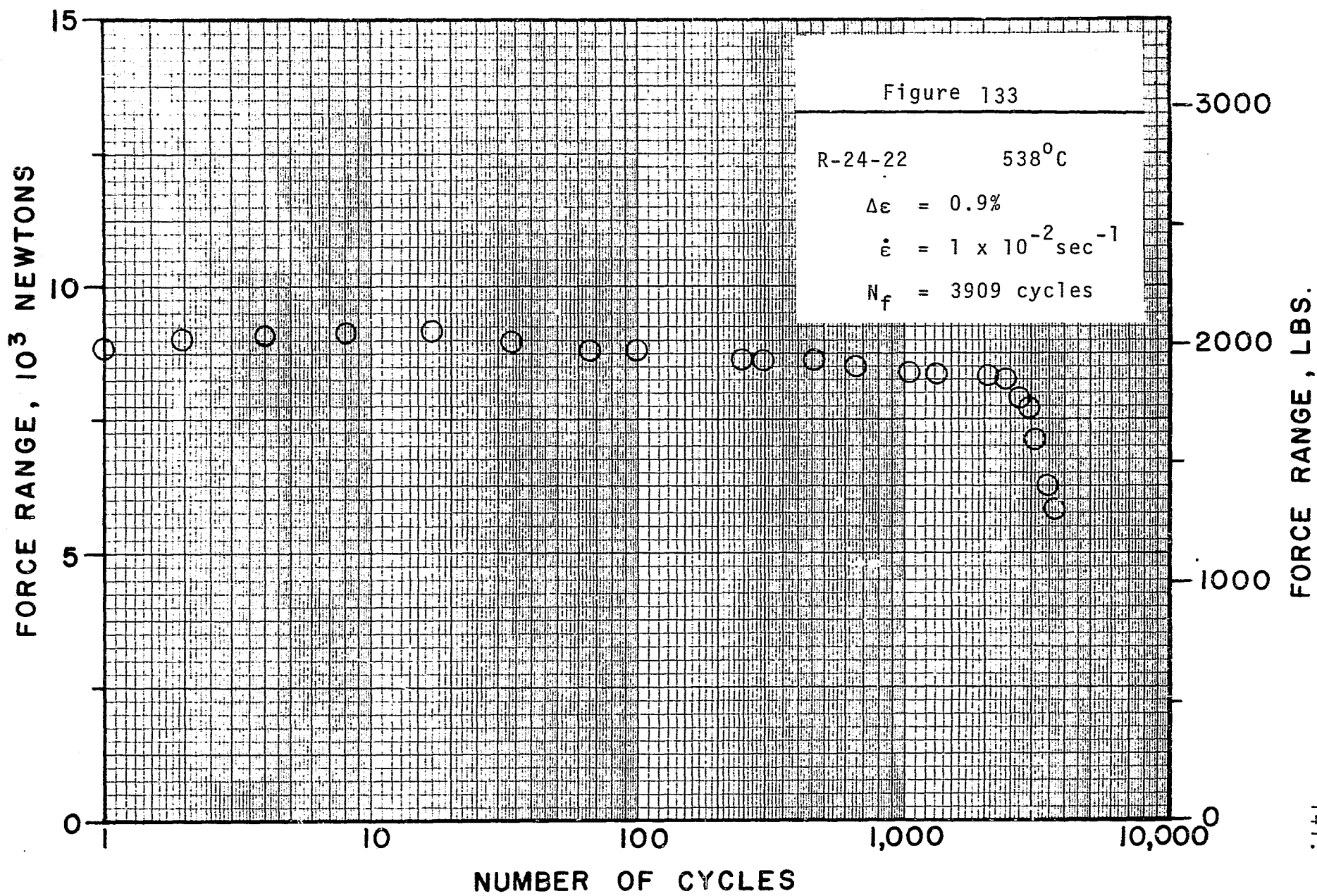


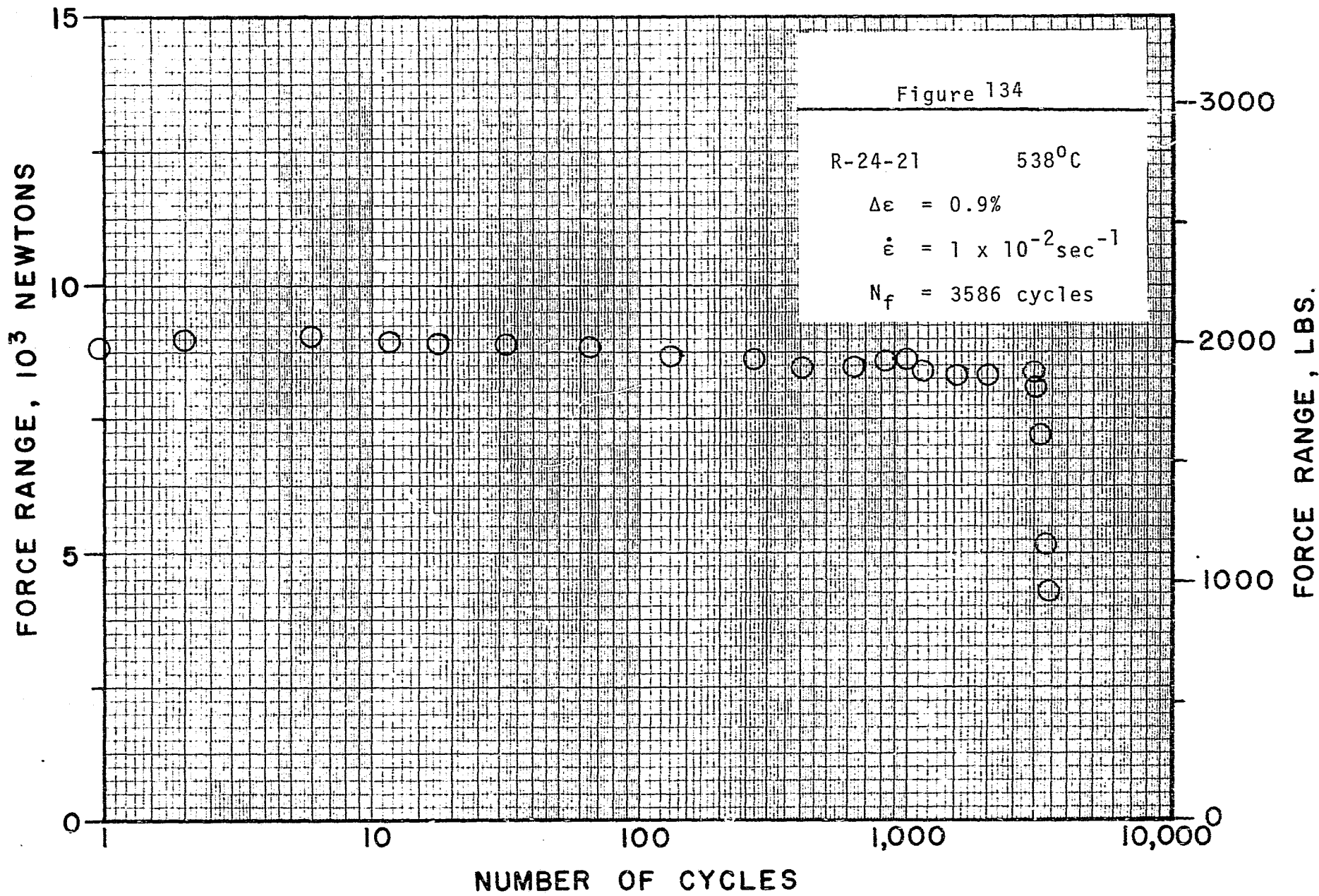


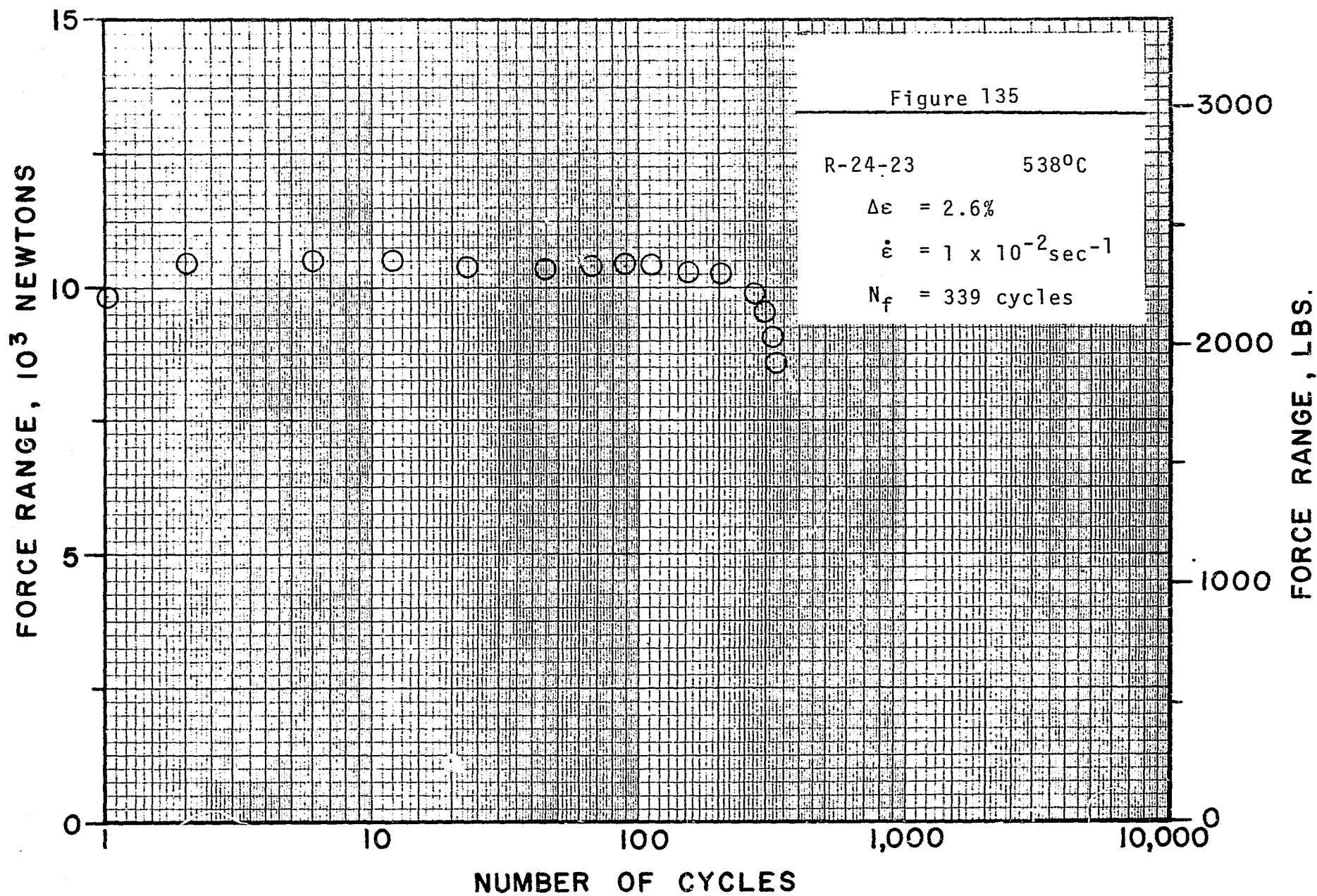


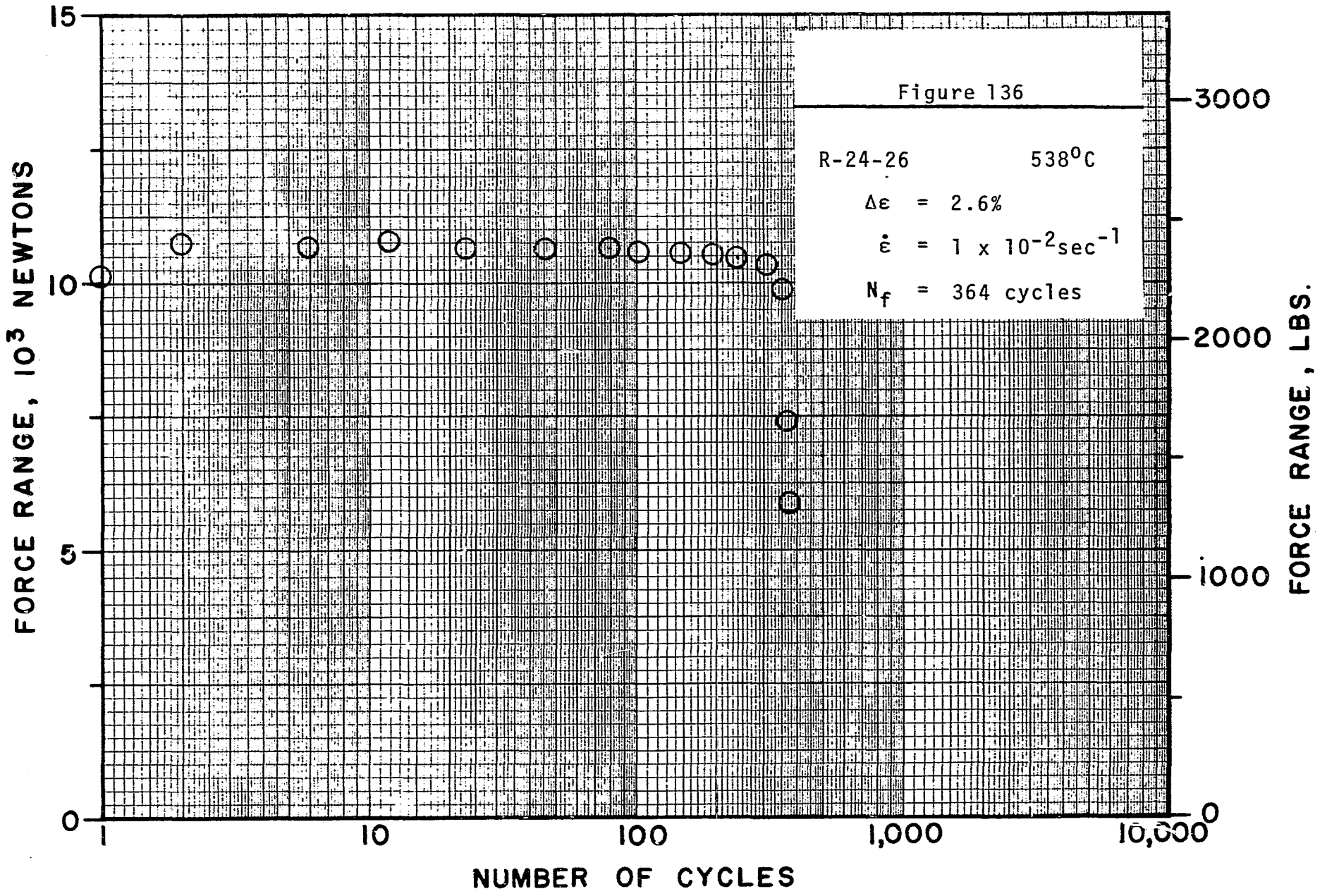


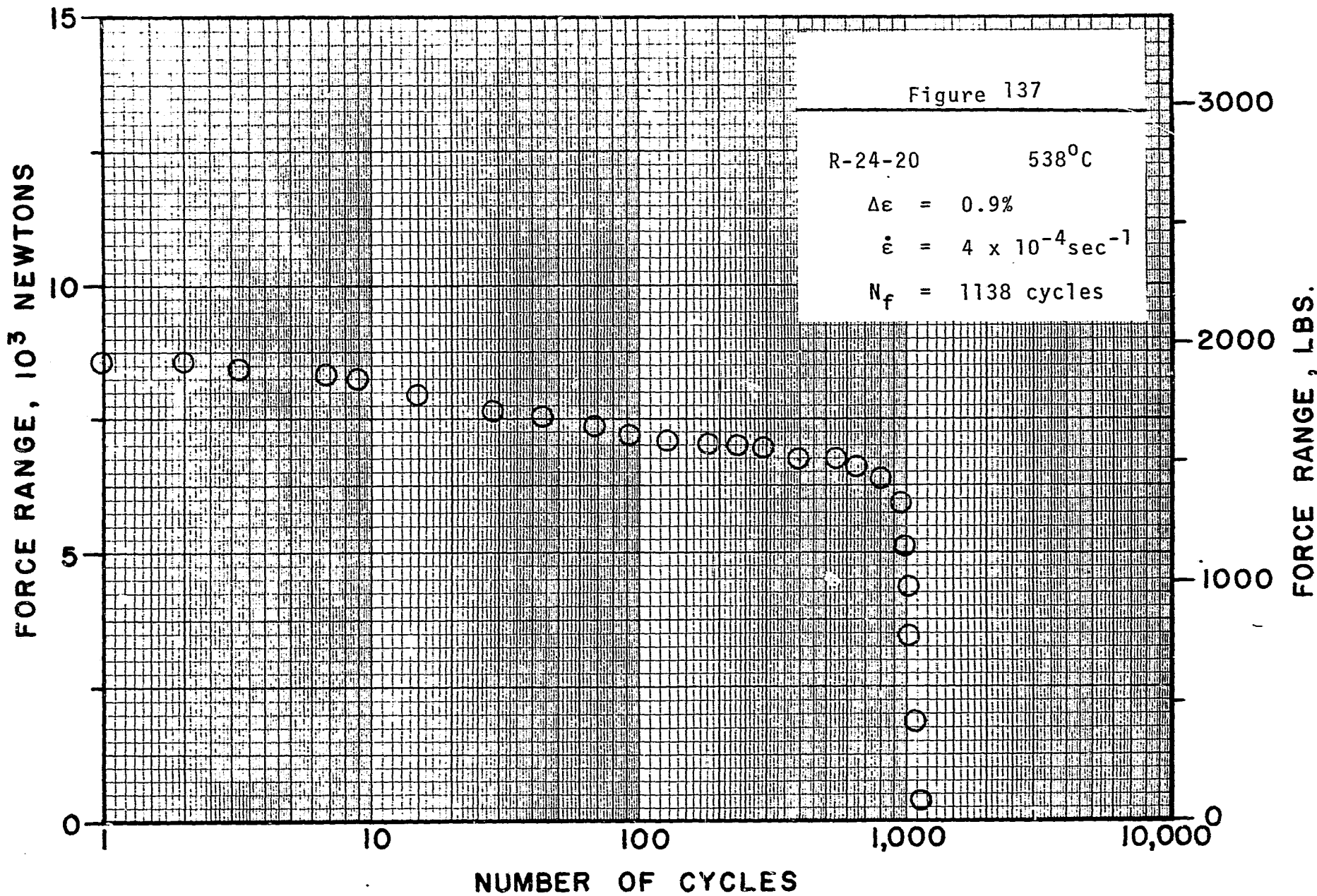


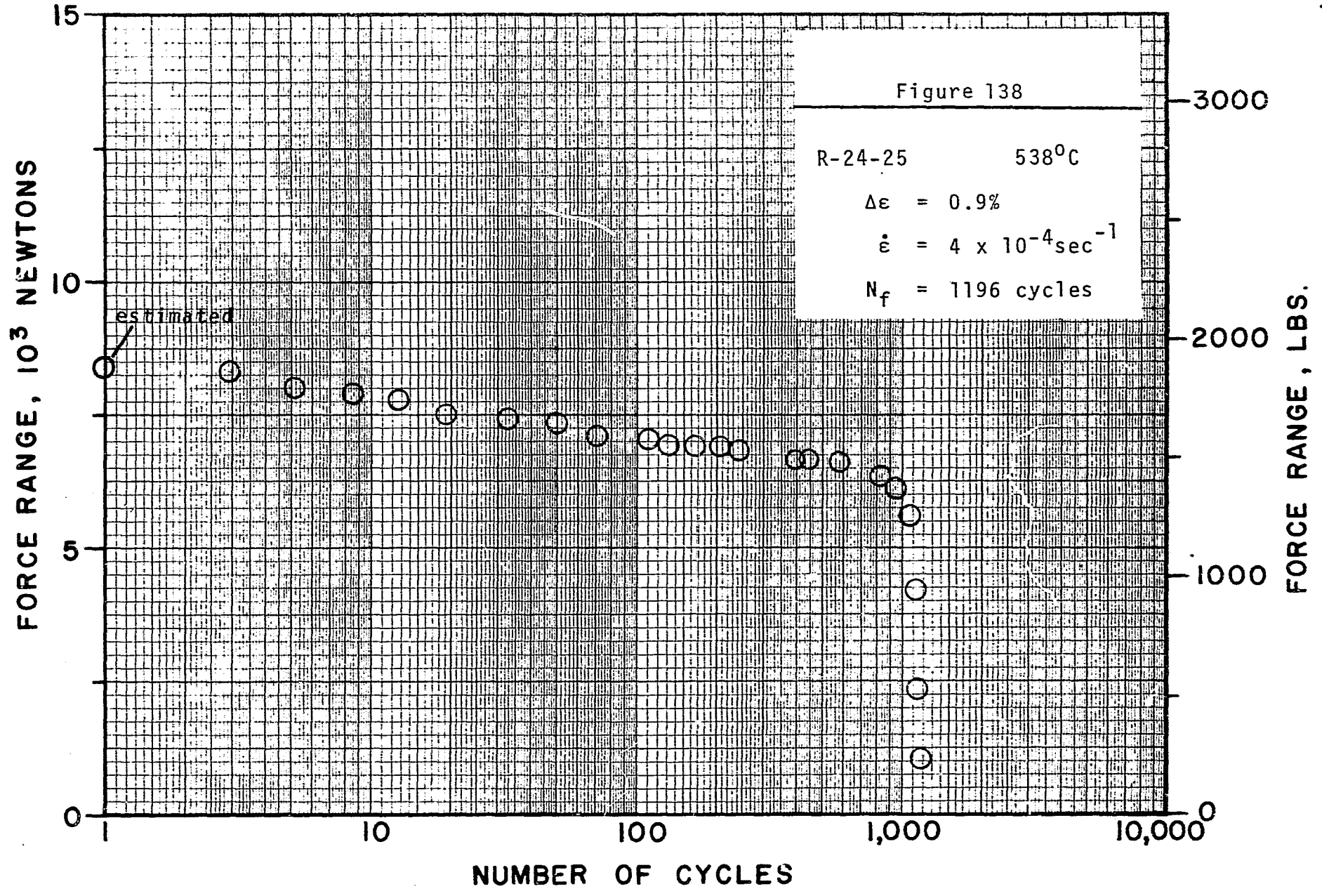


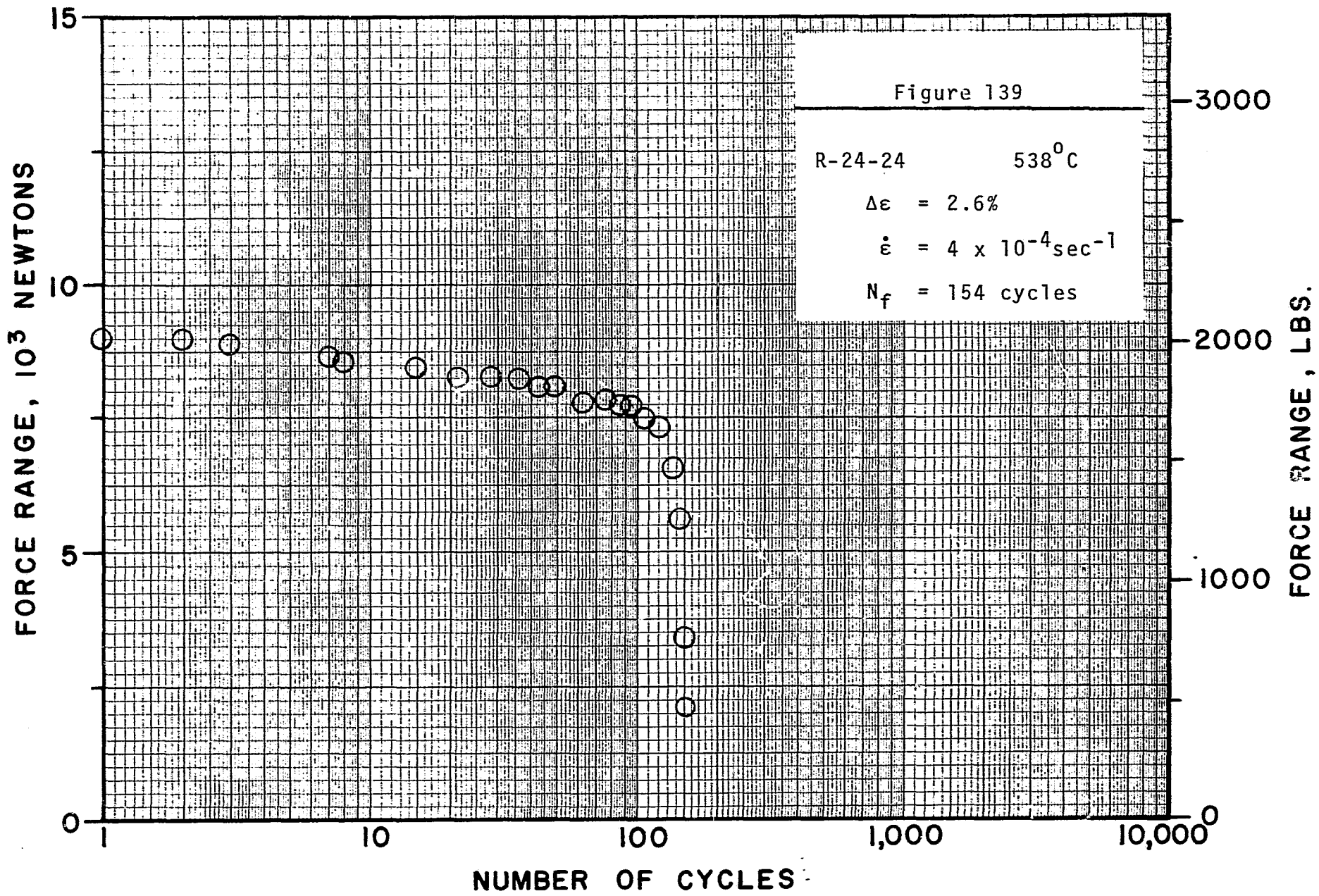


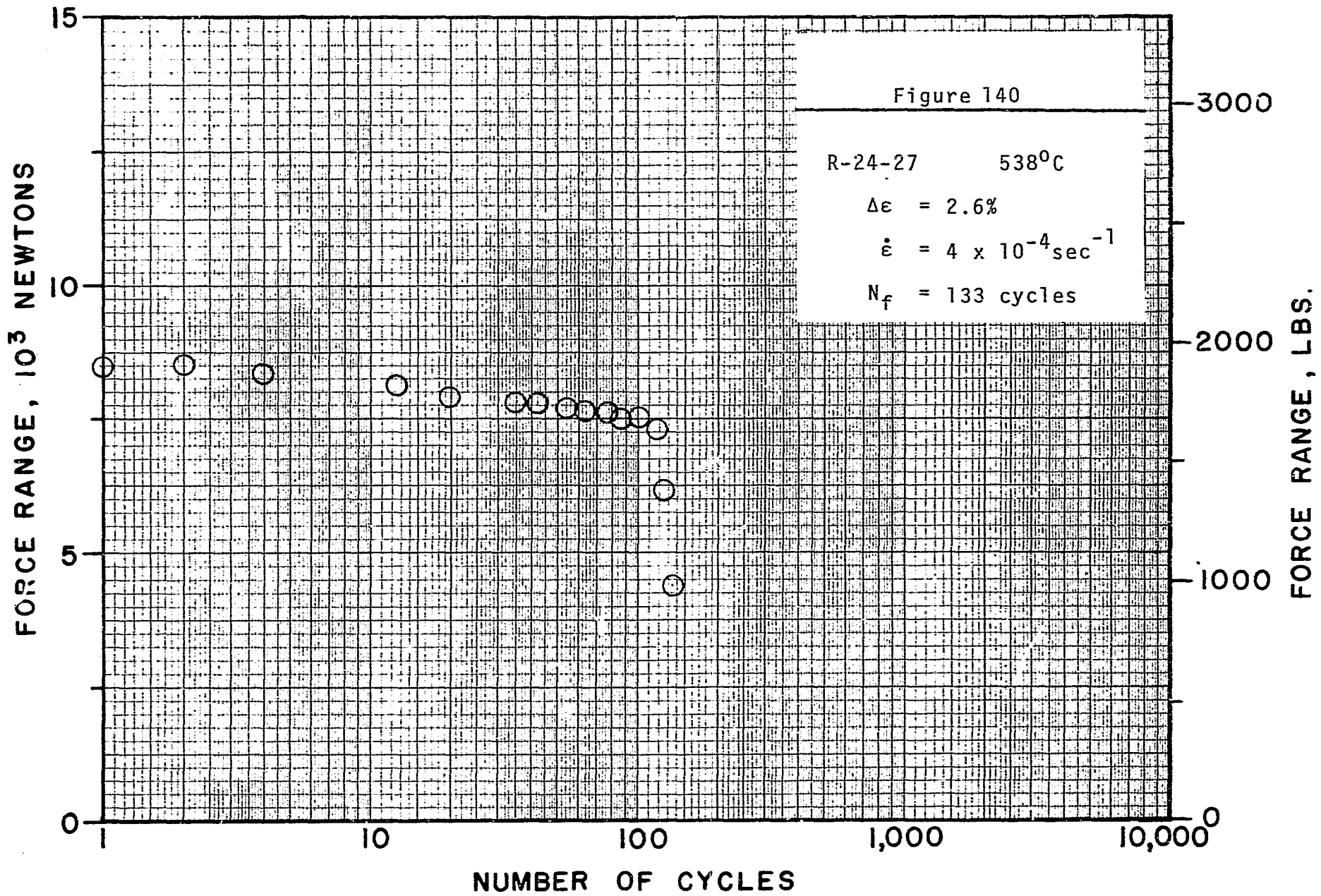


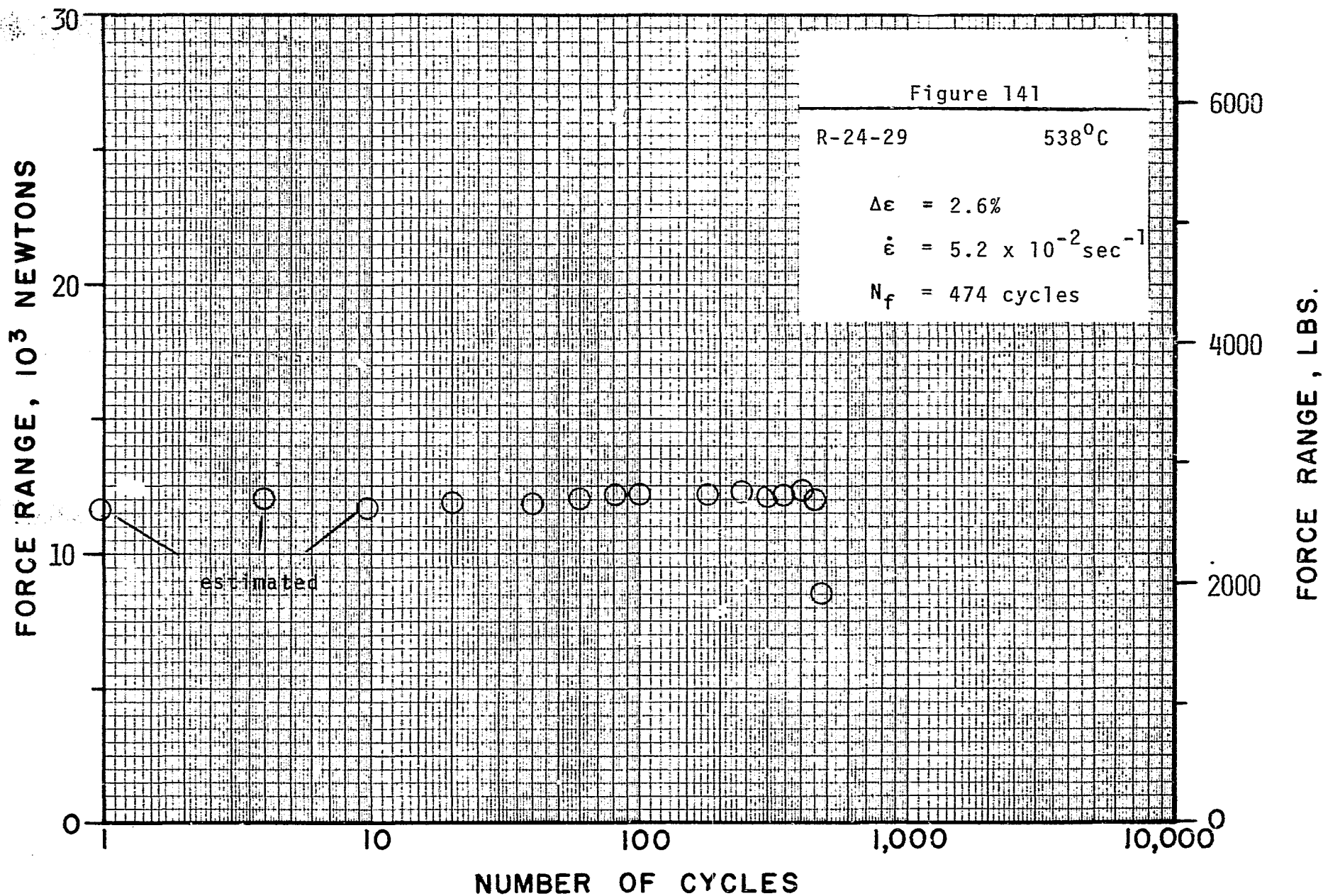


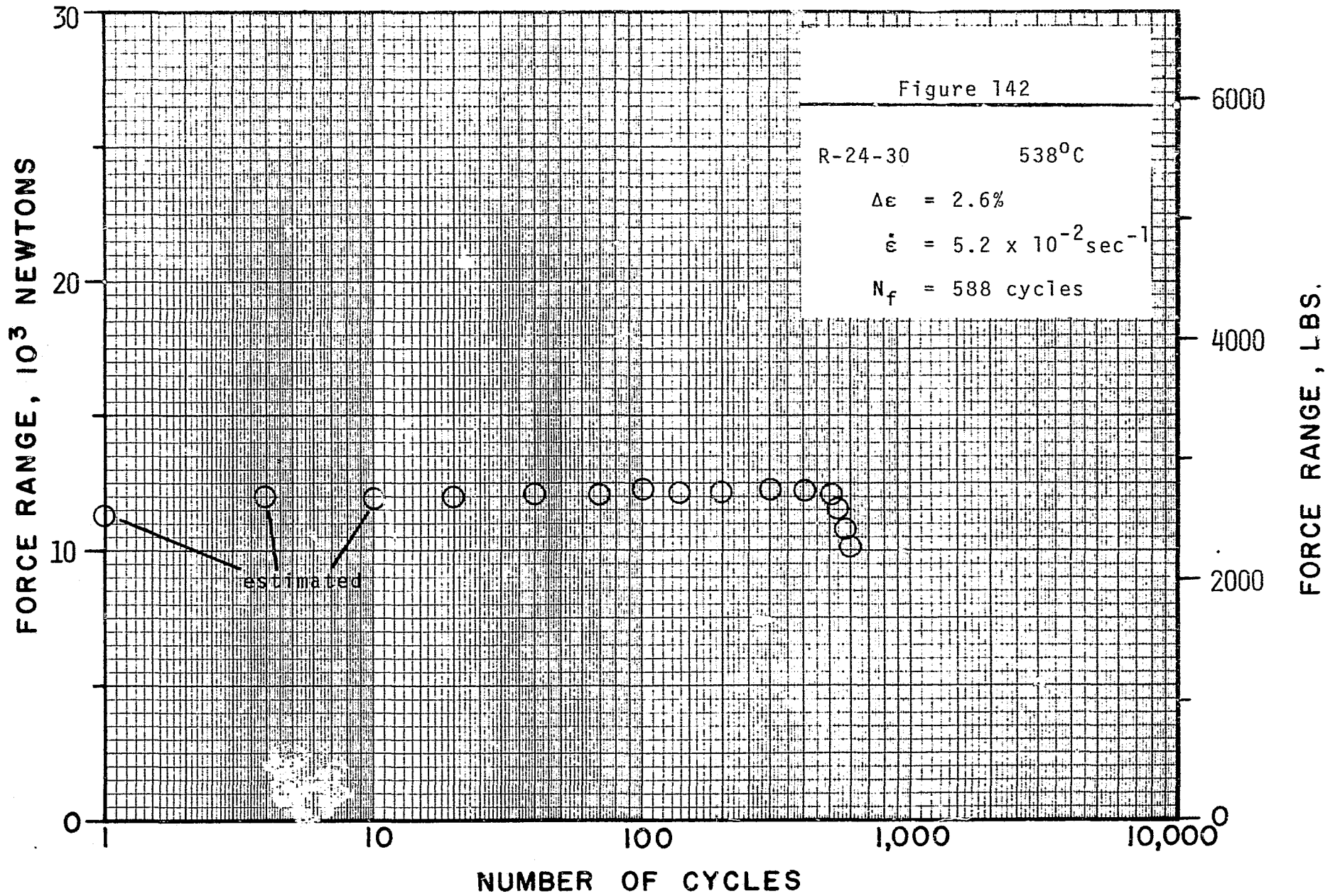


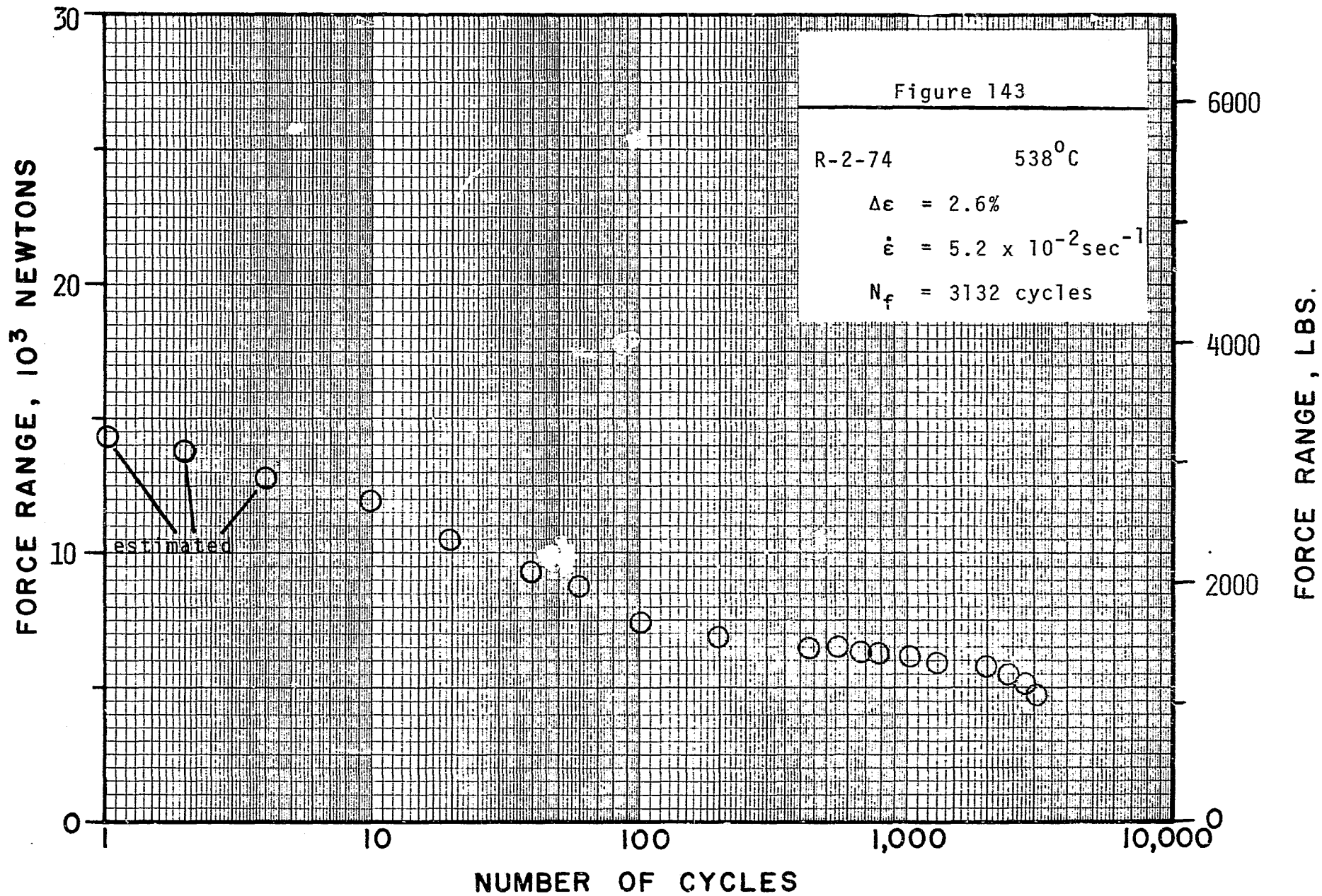


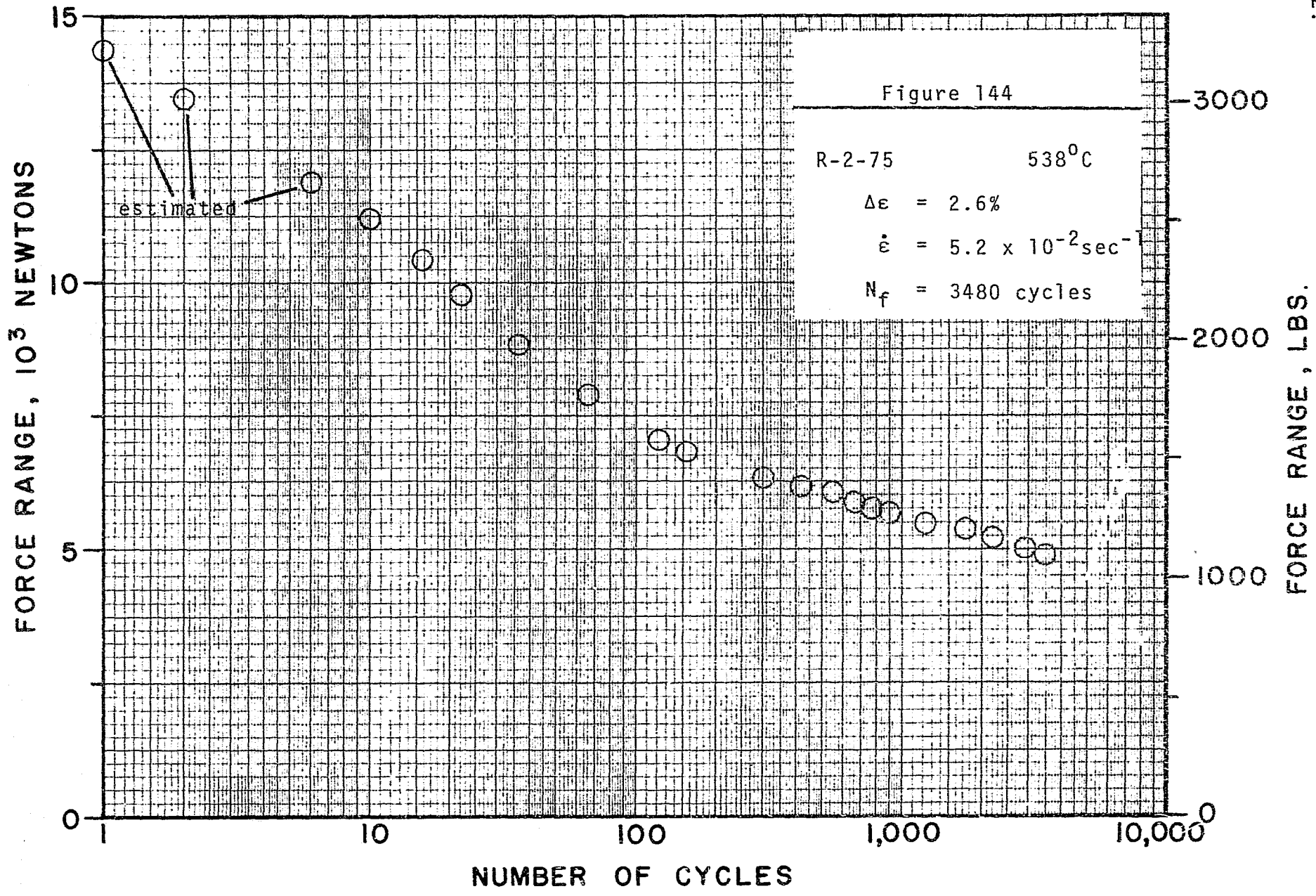


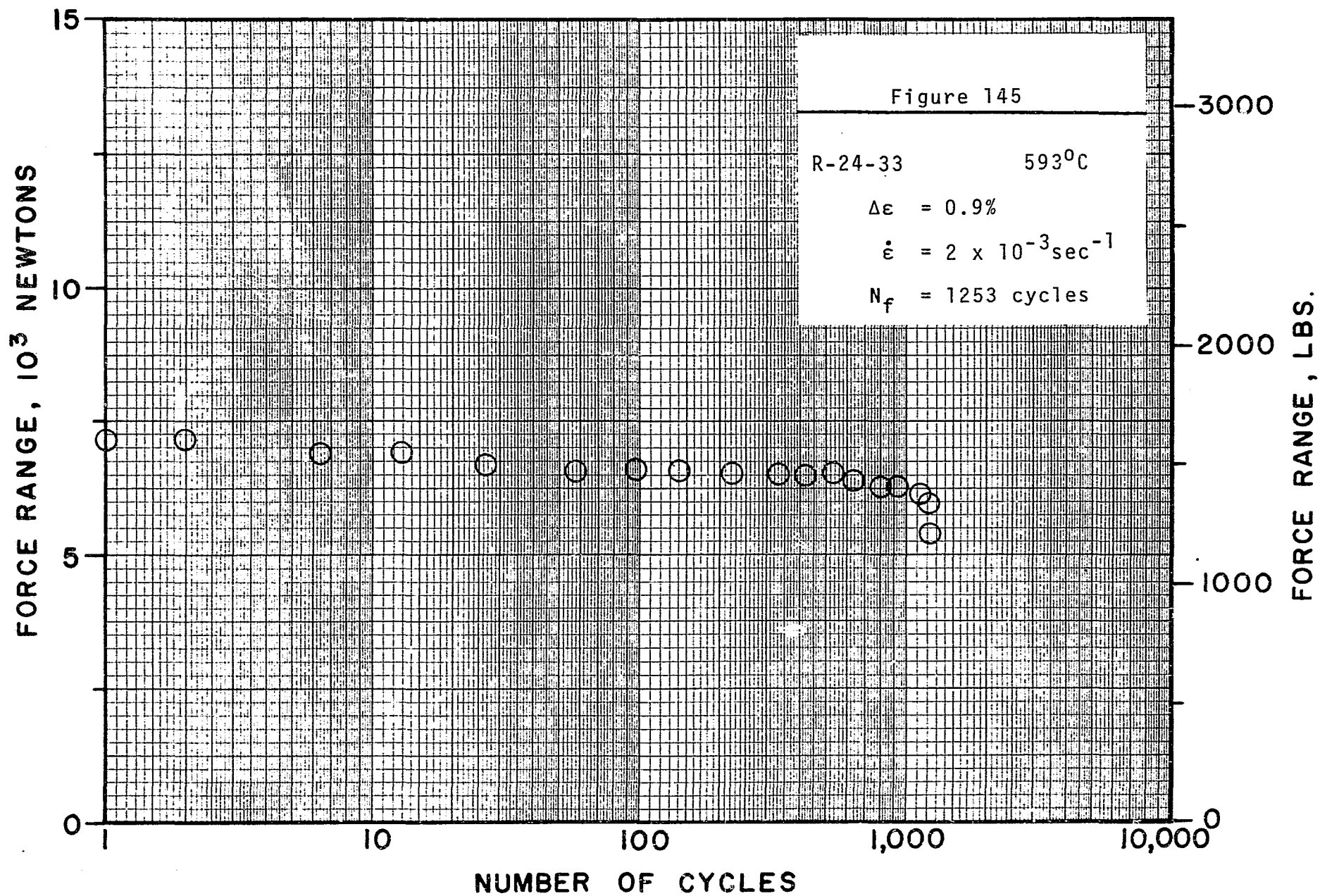


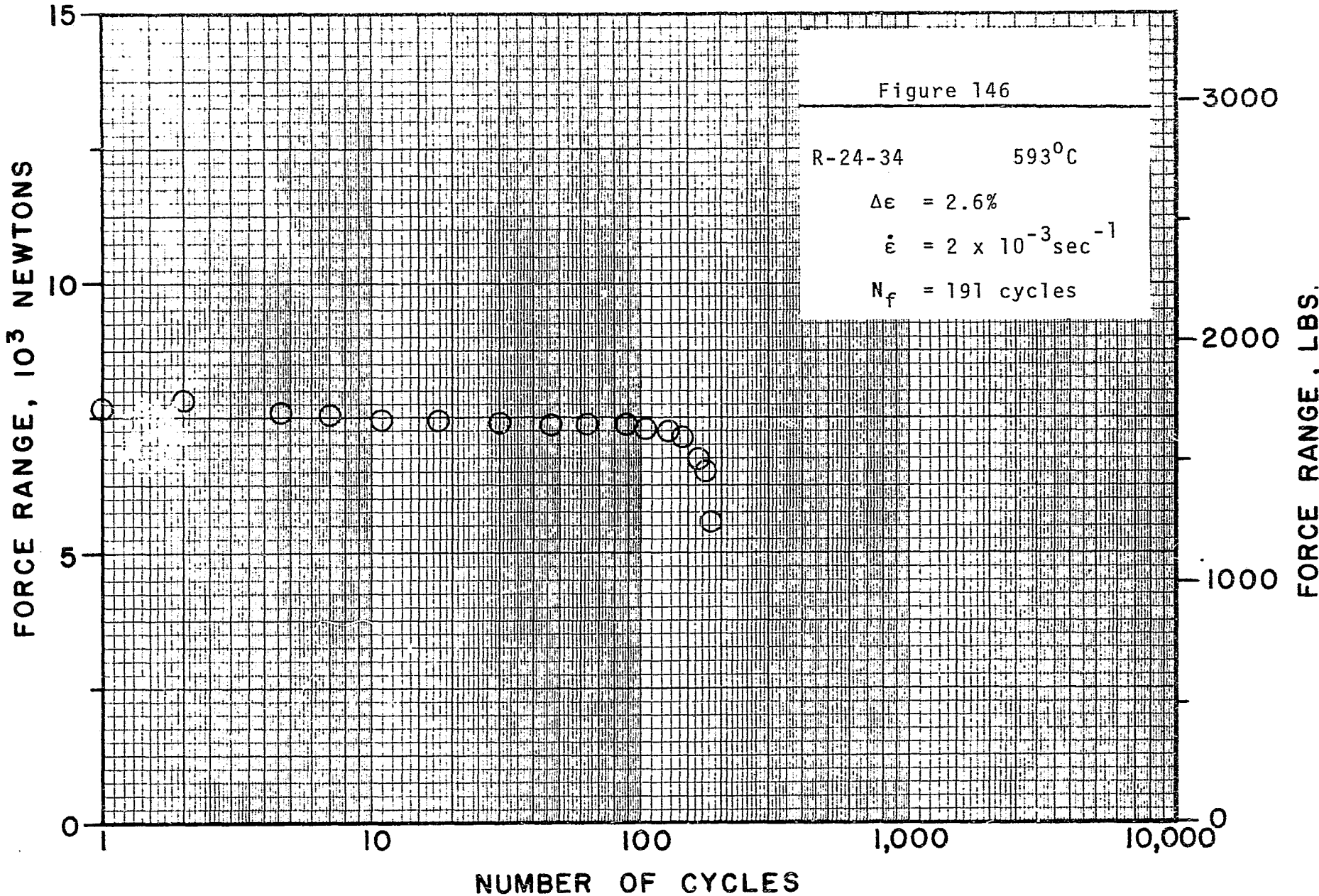


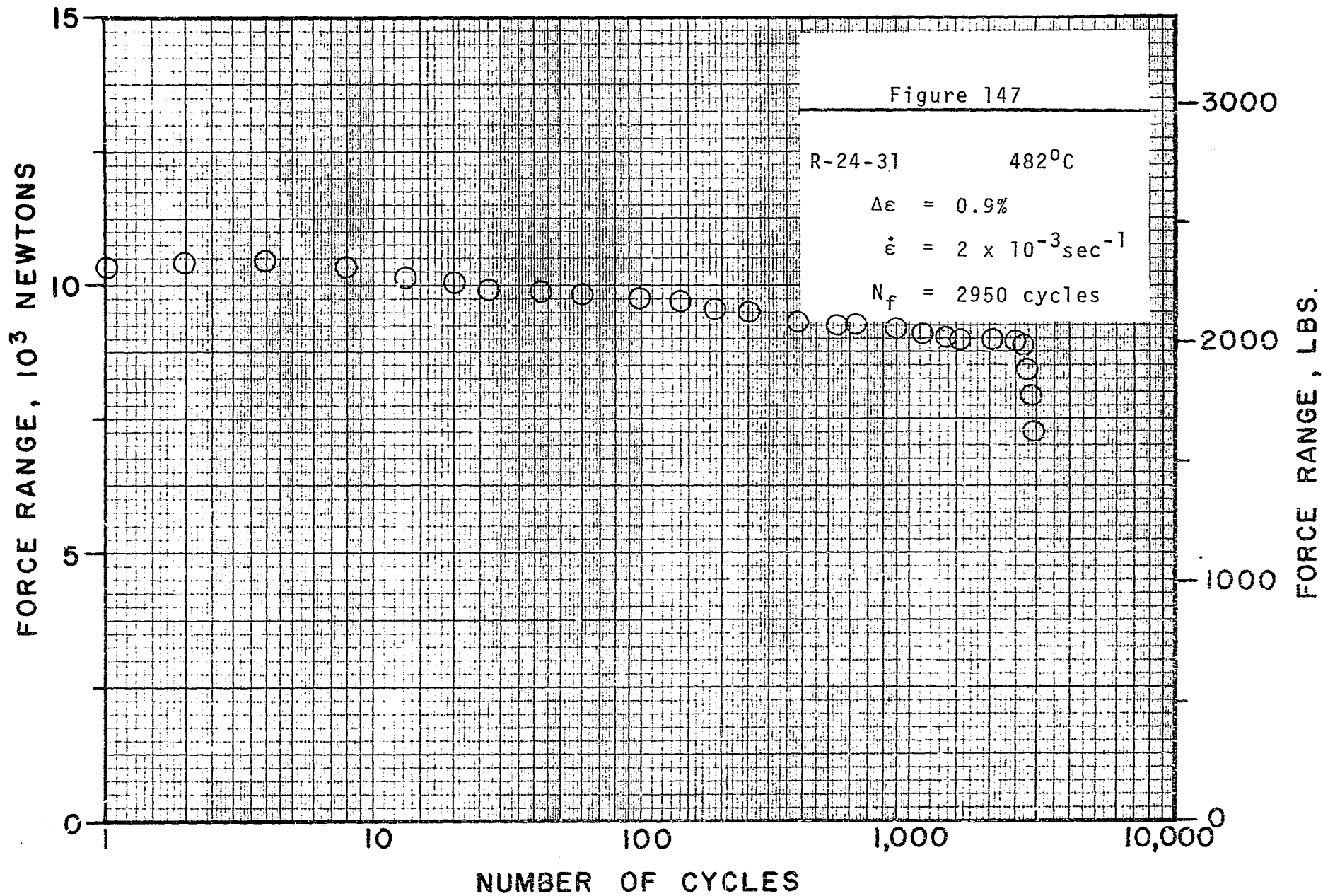


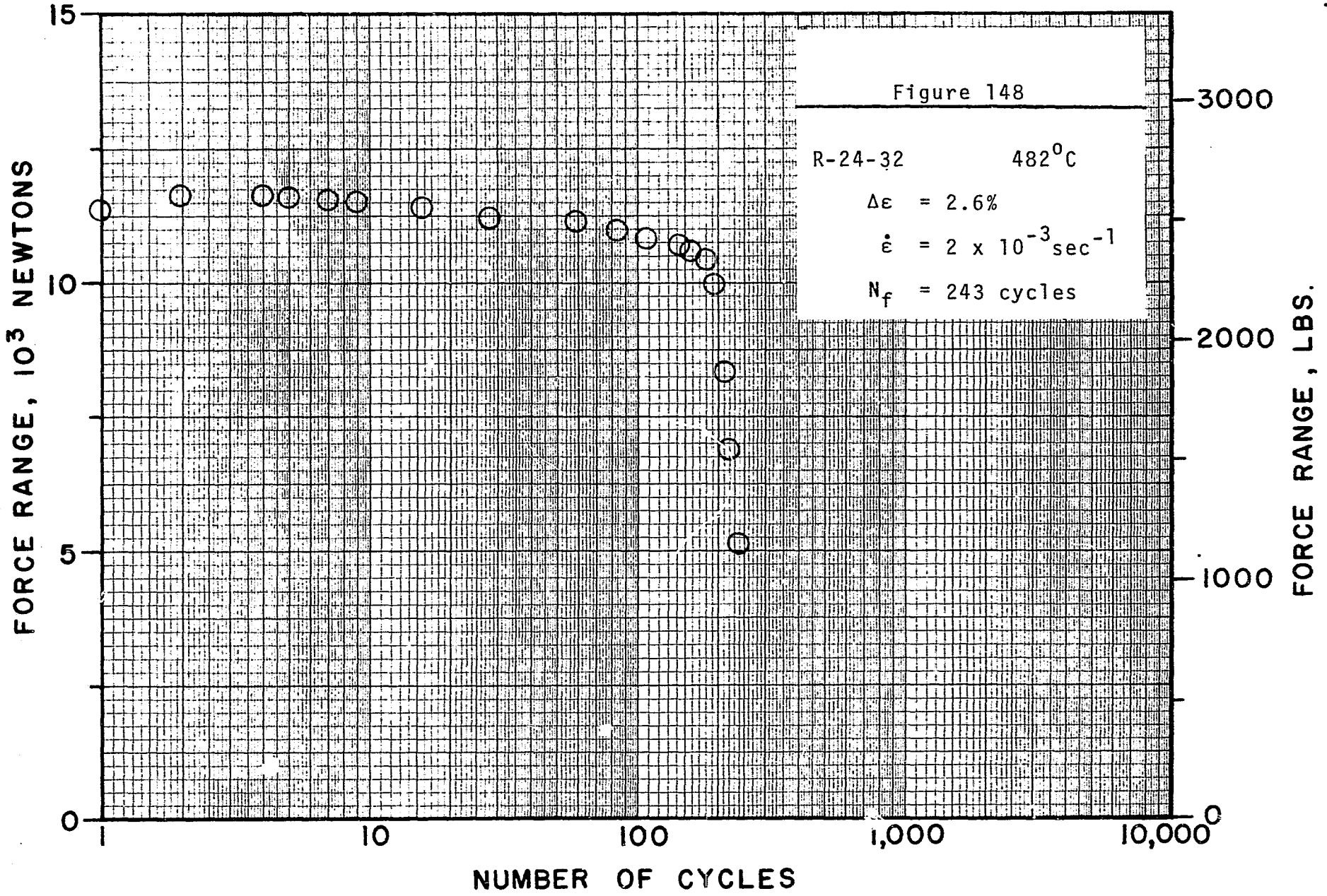


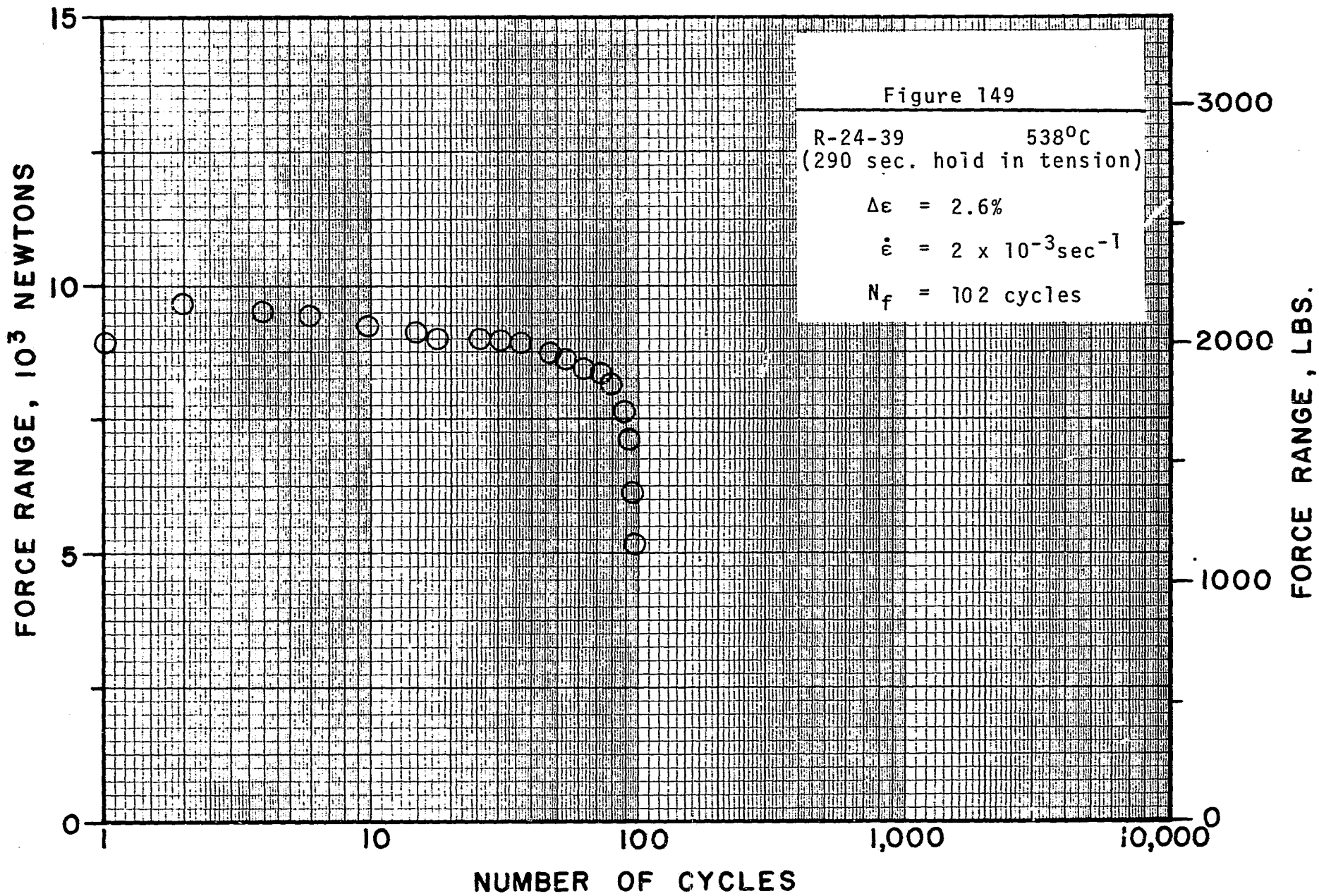


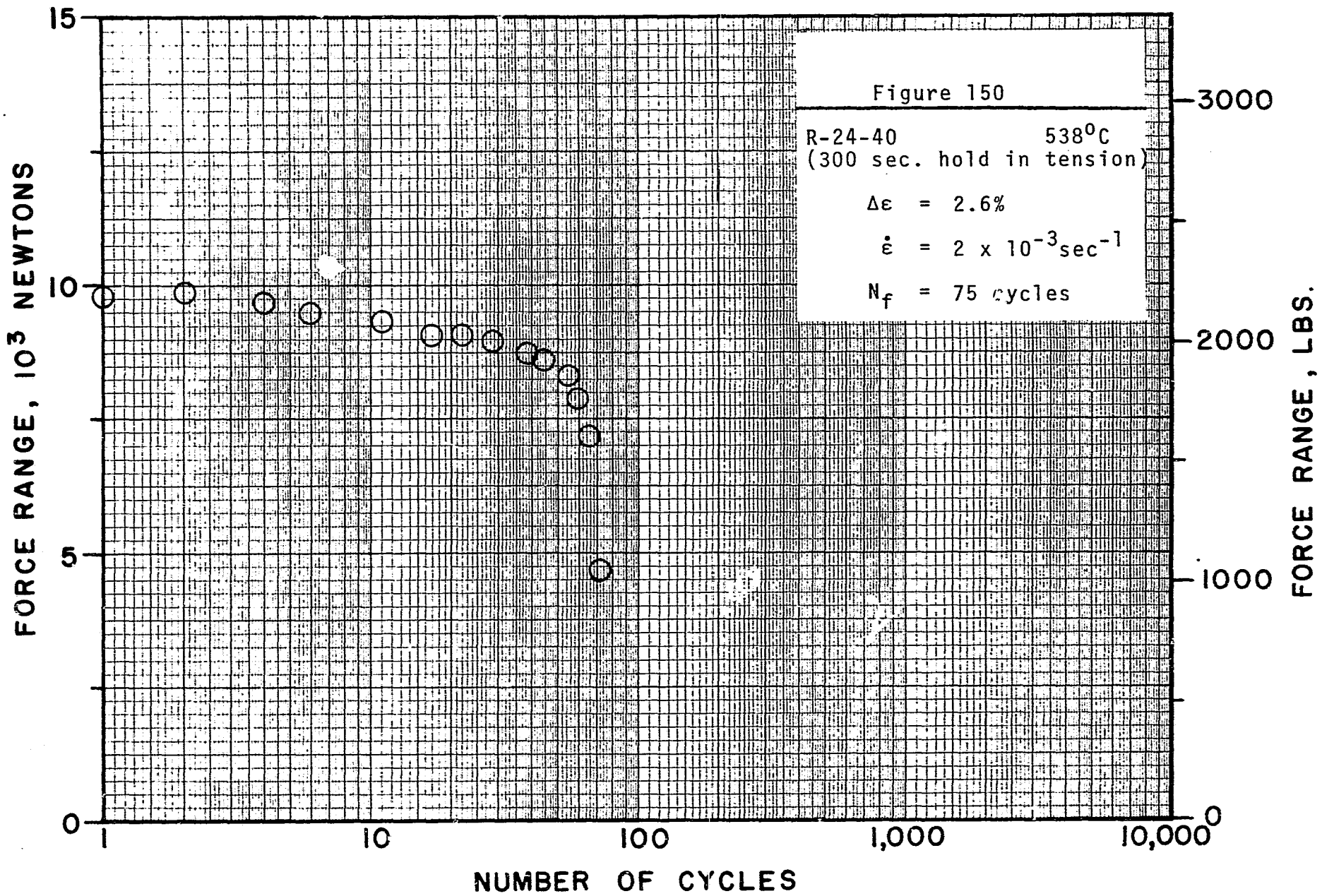


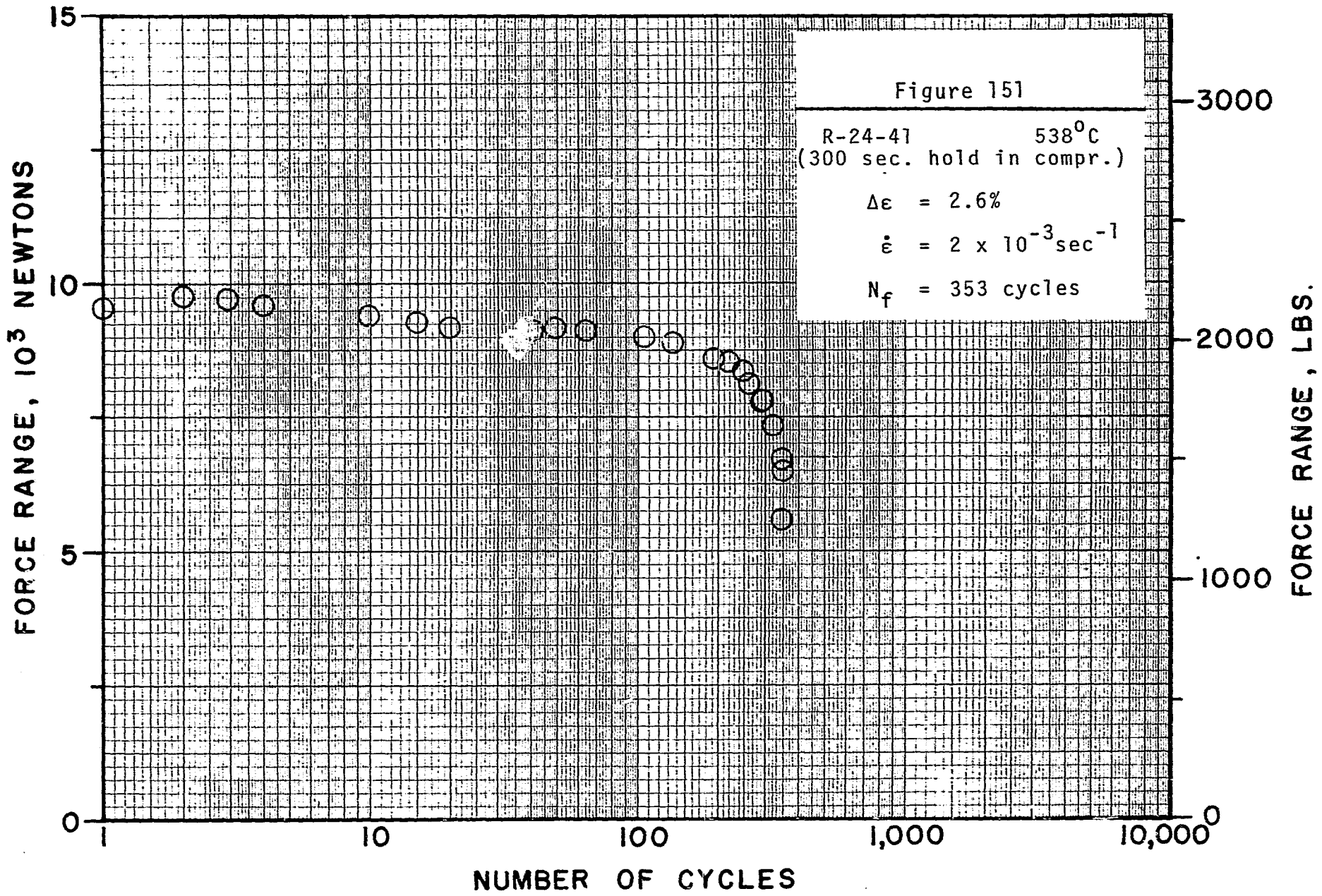


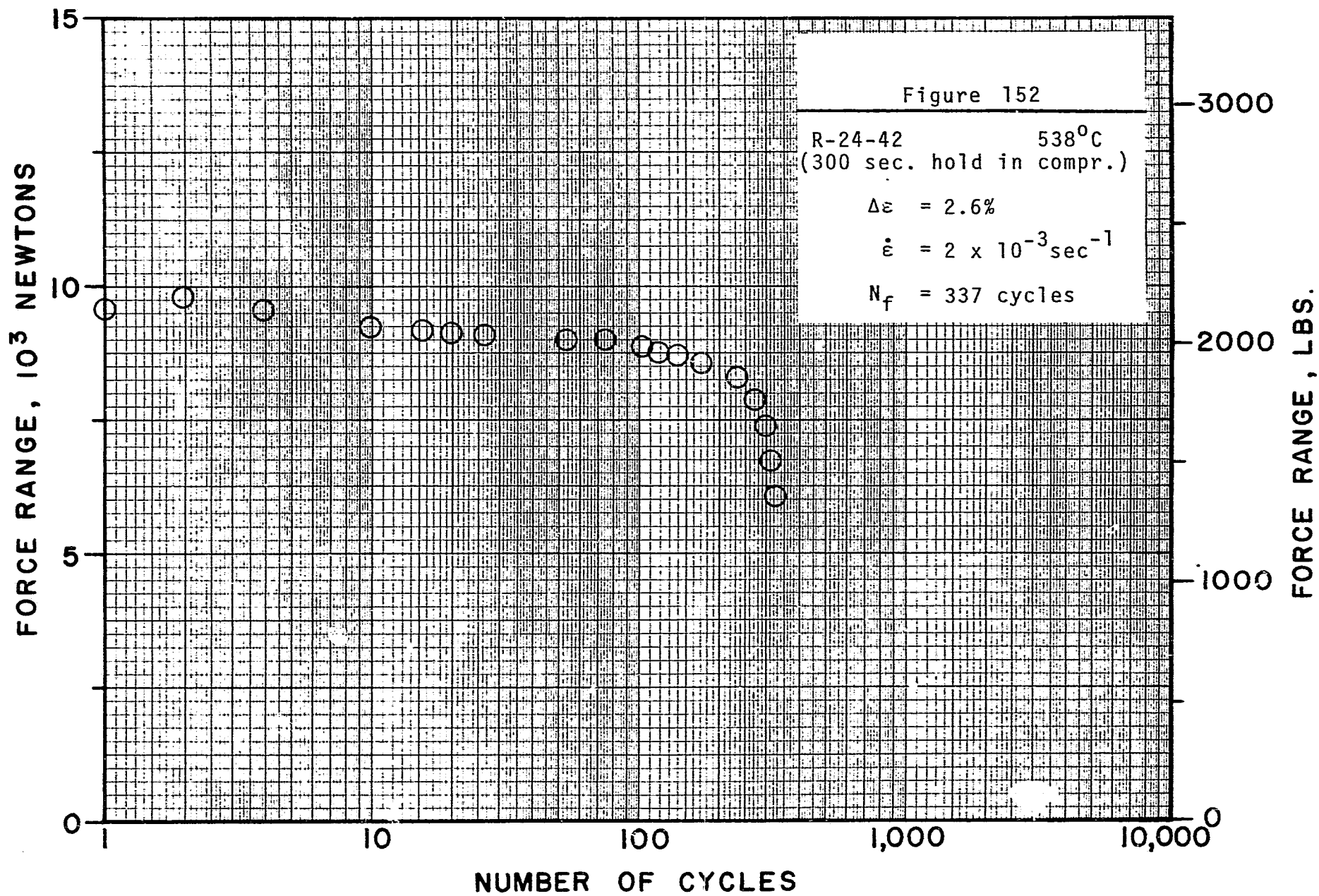


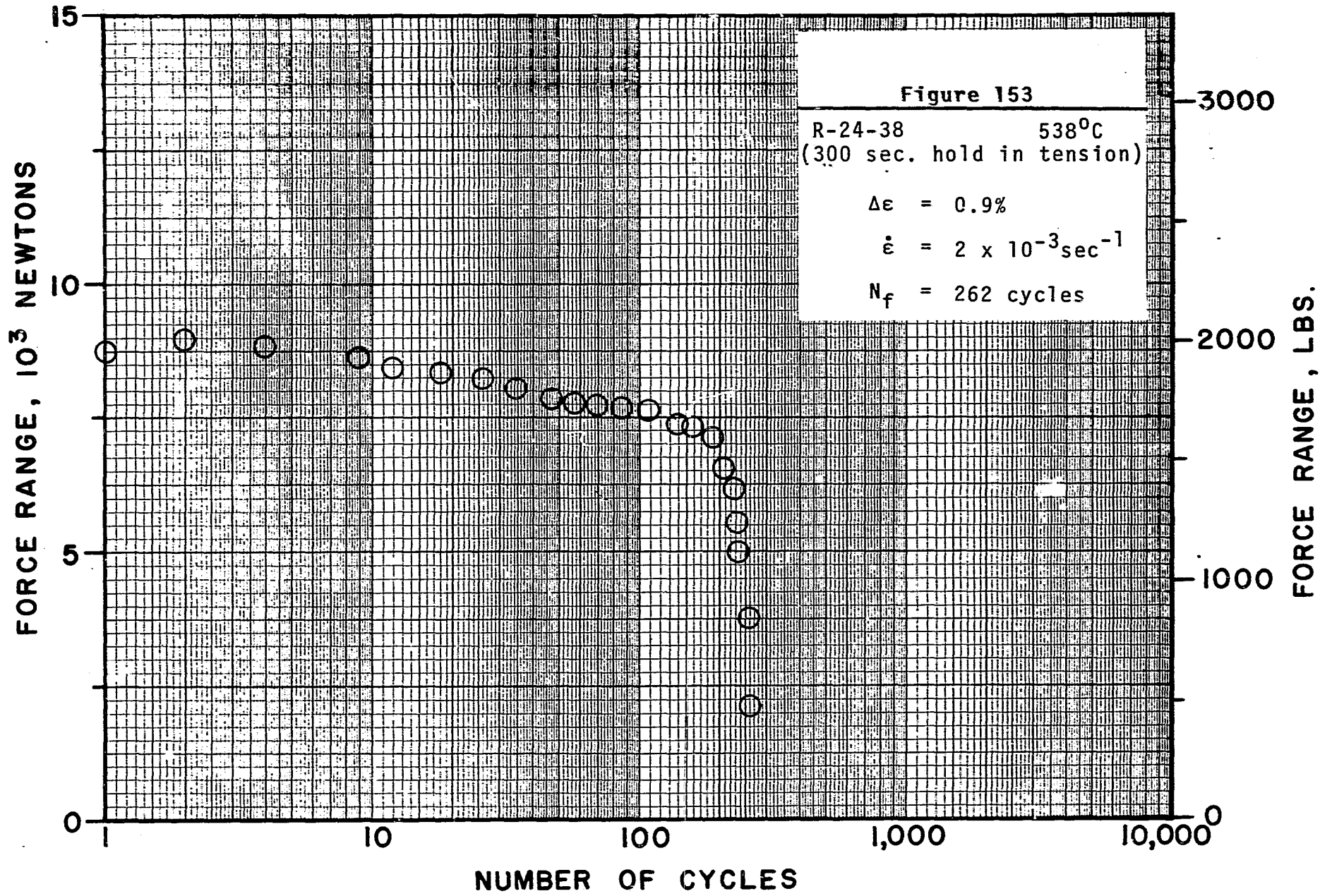


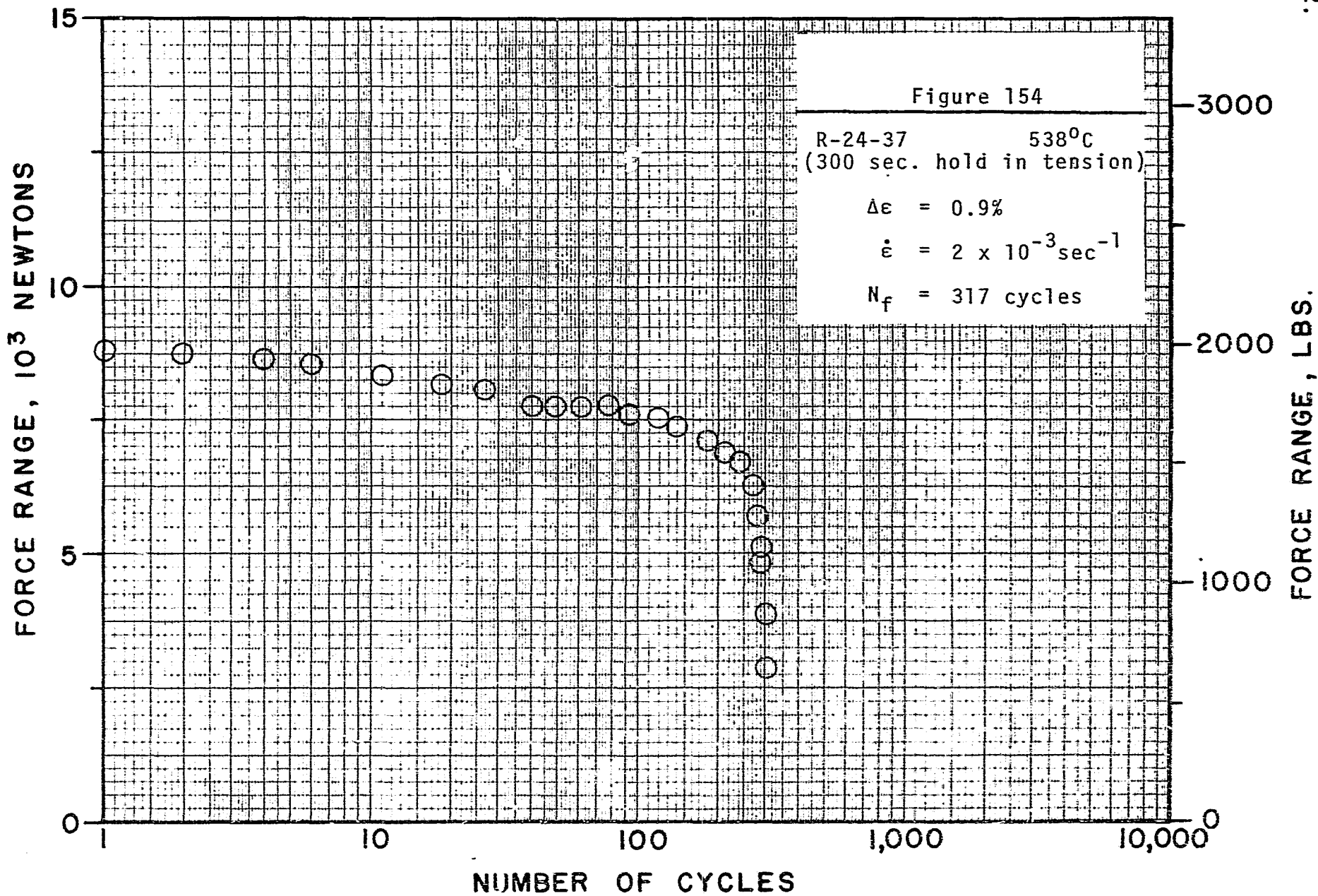




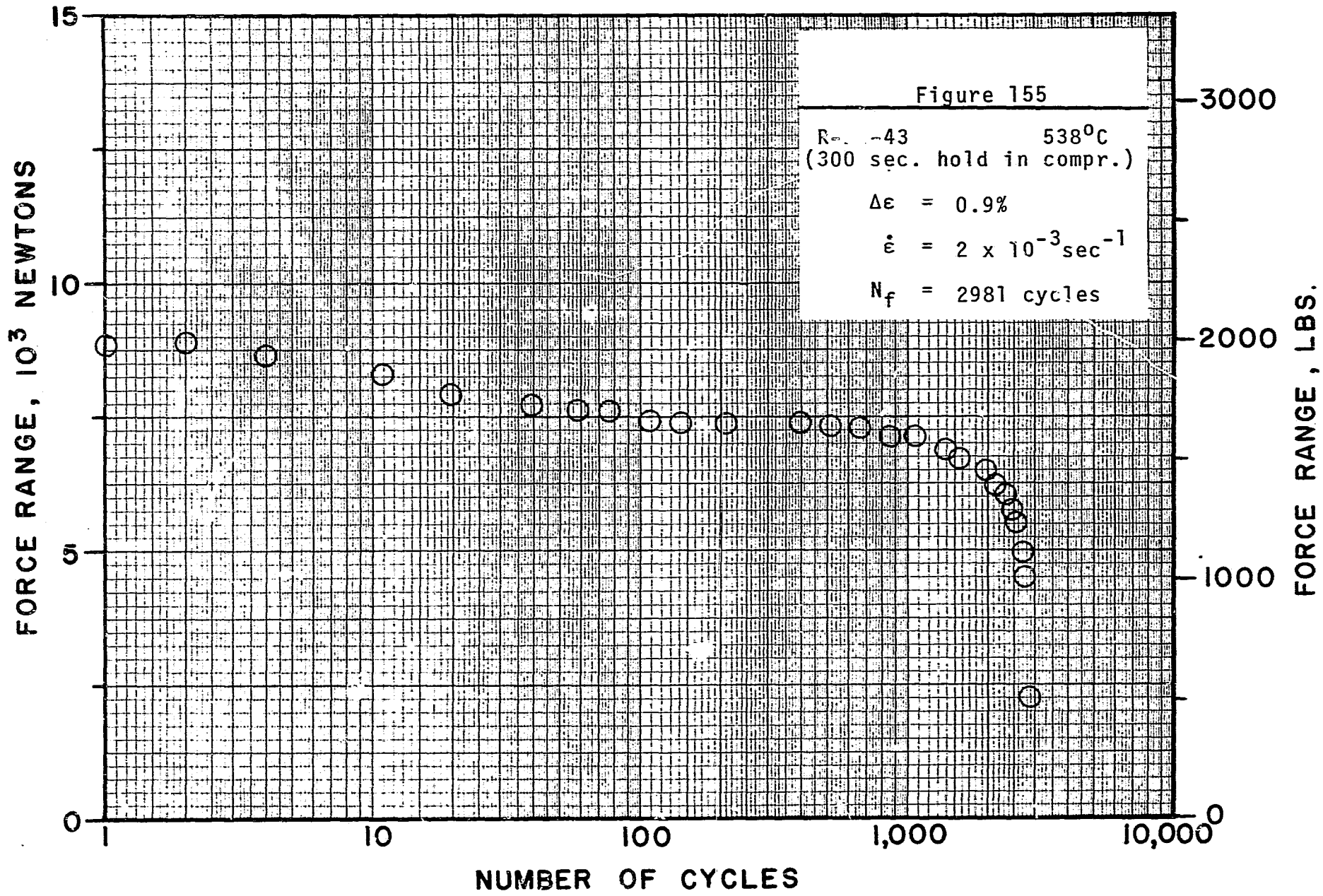


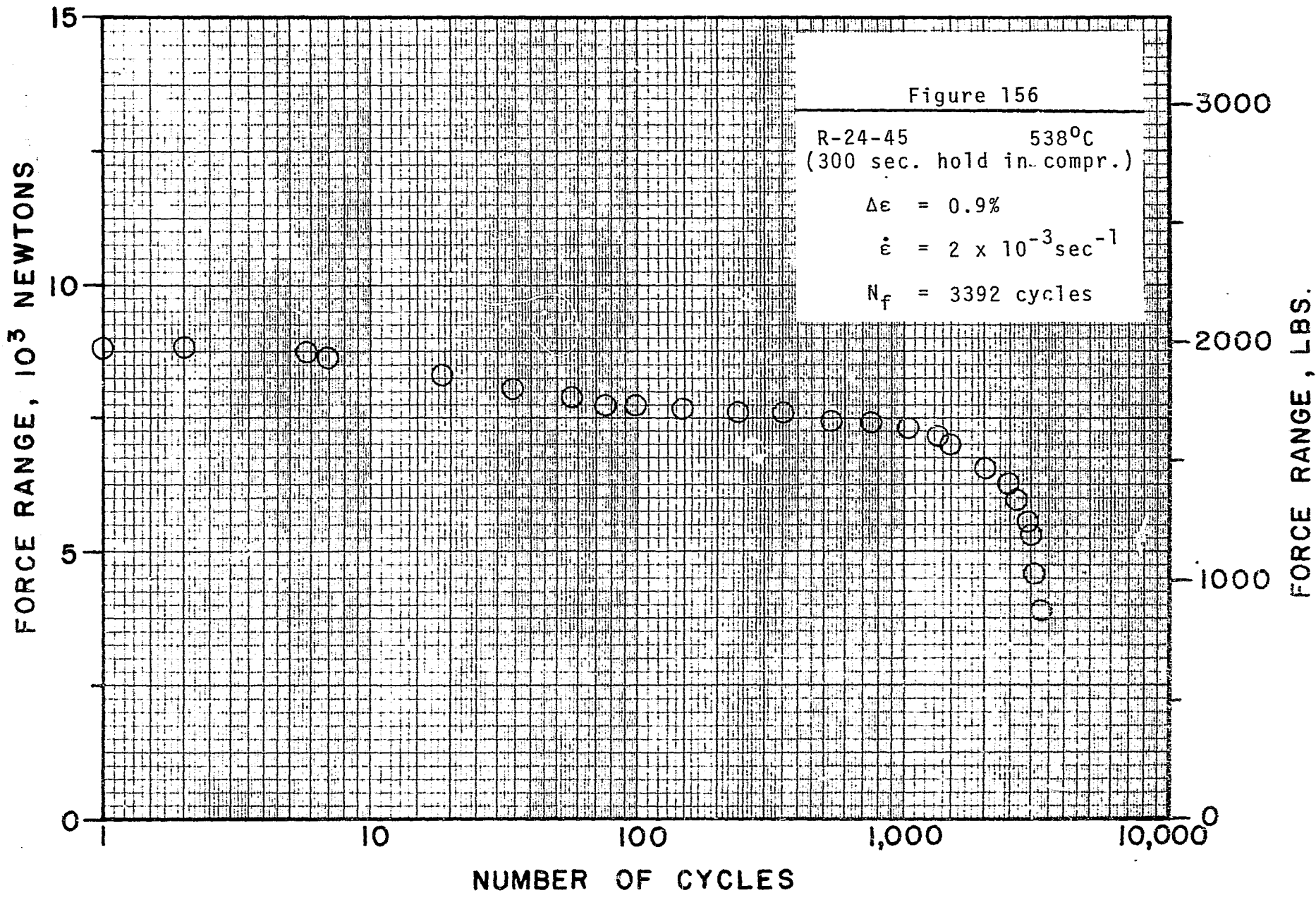


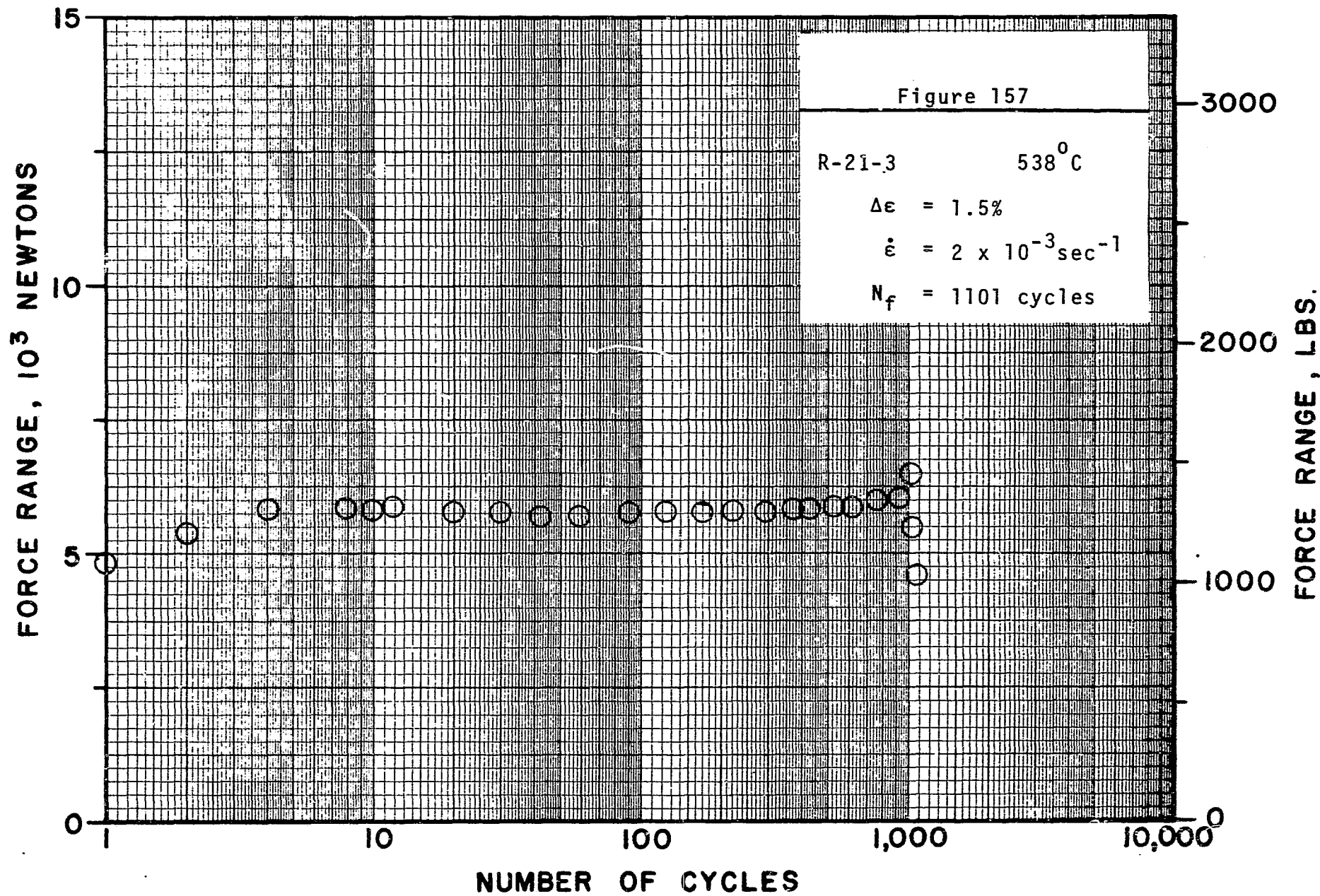


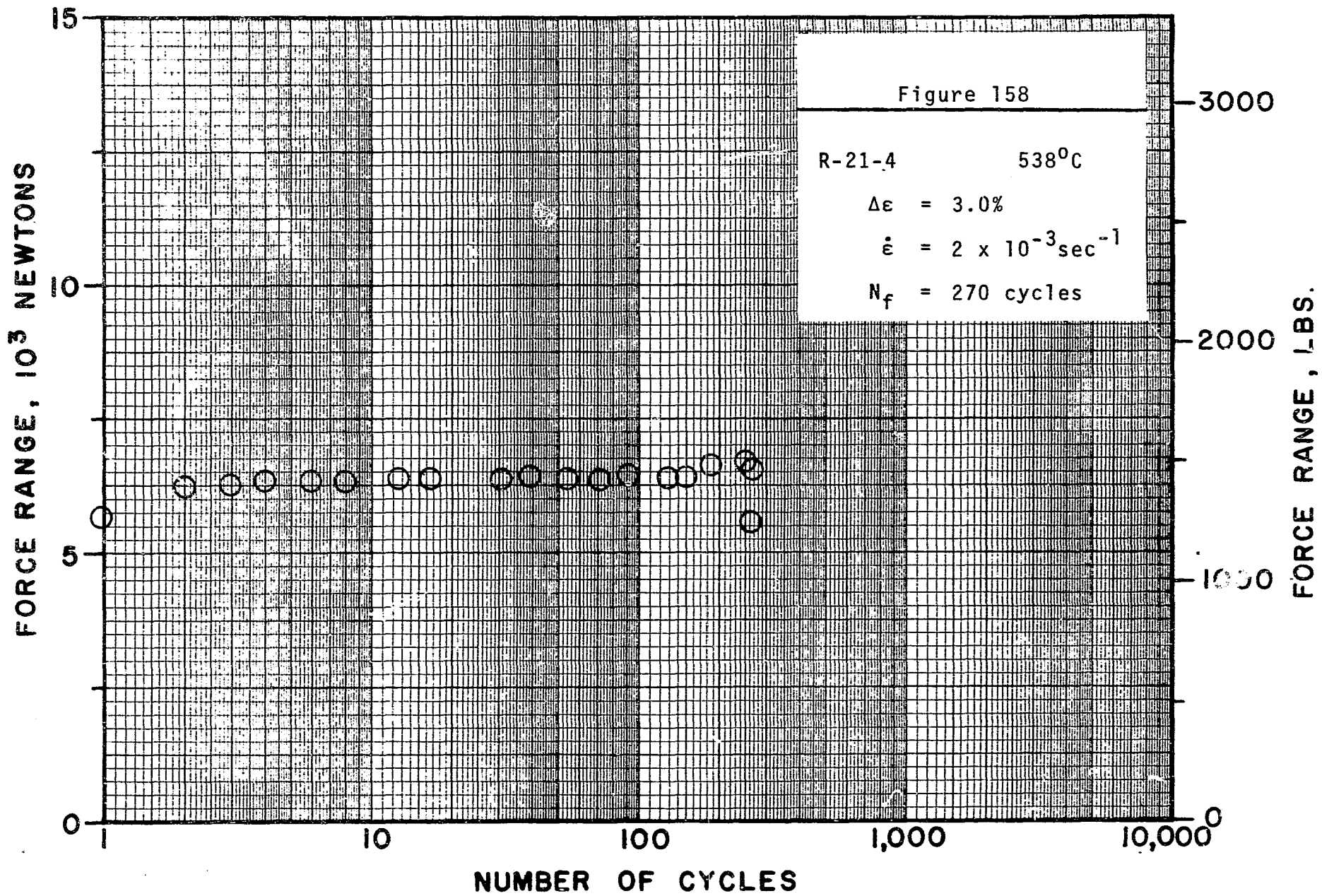


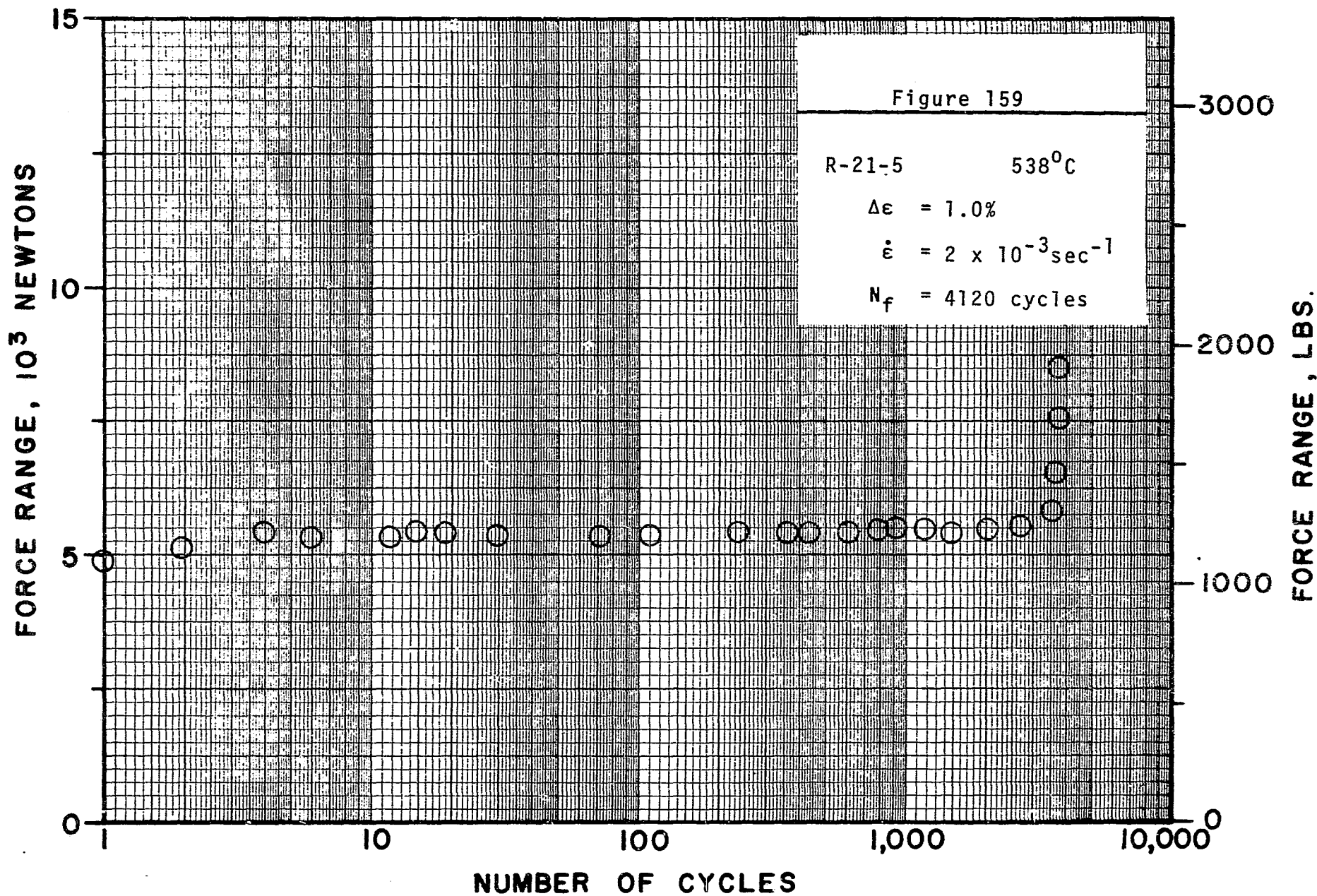
FORCE RANGE, LBS.

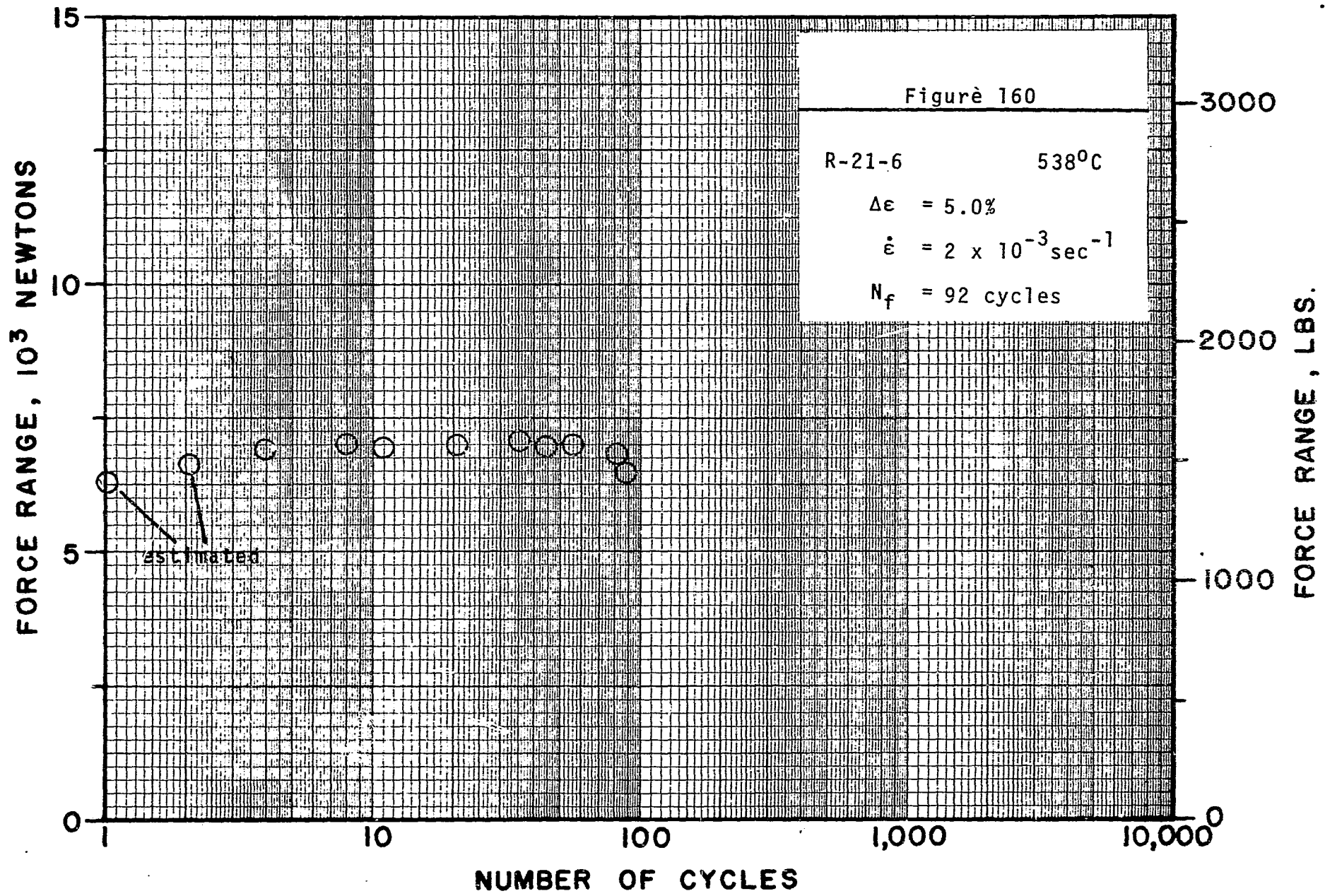


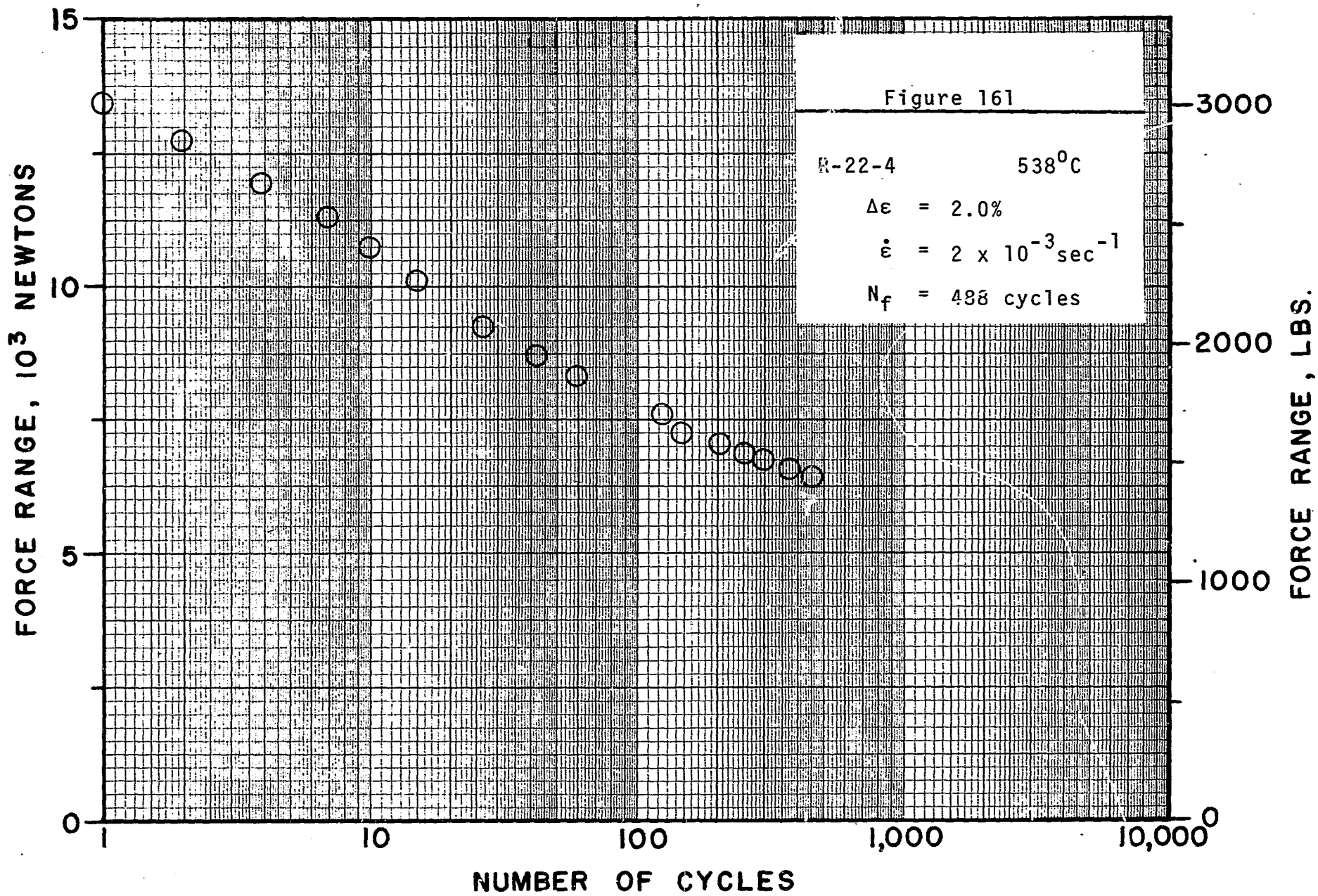


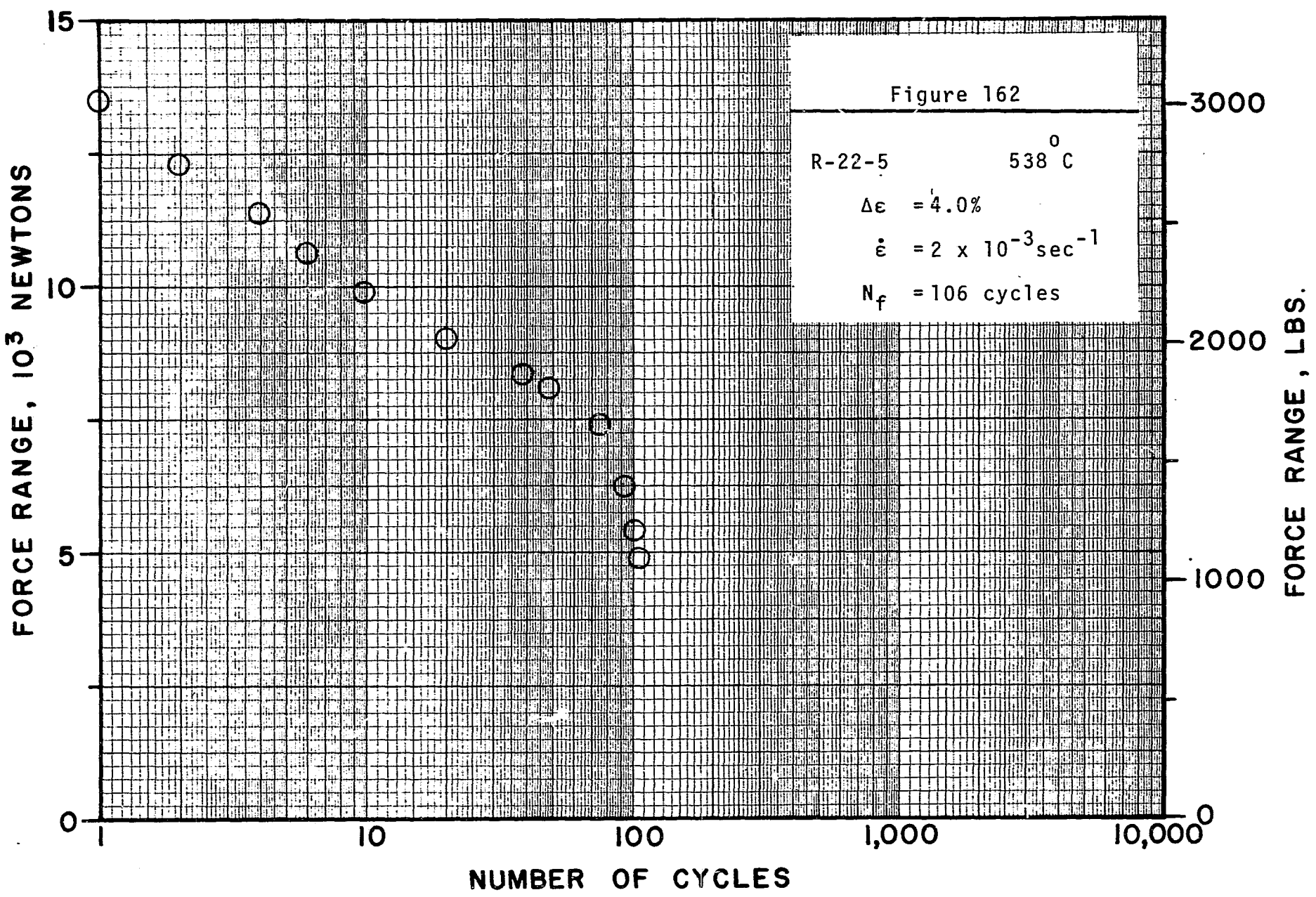


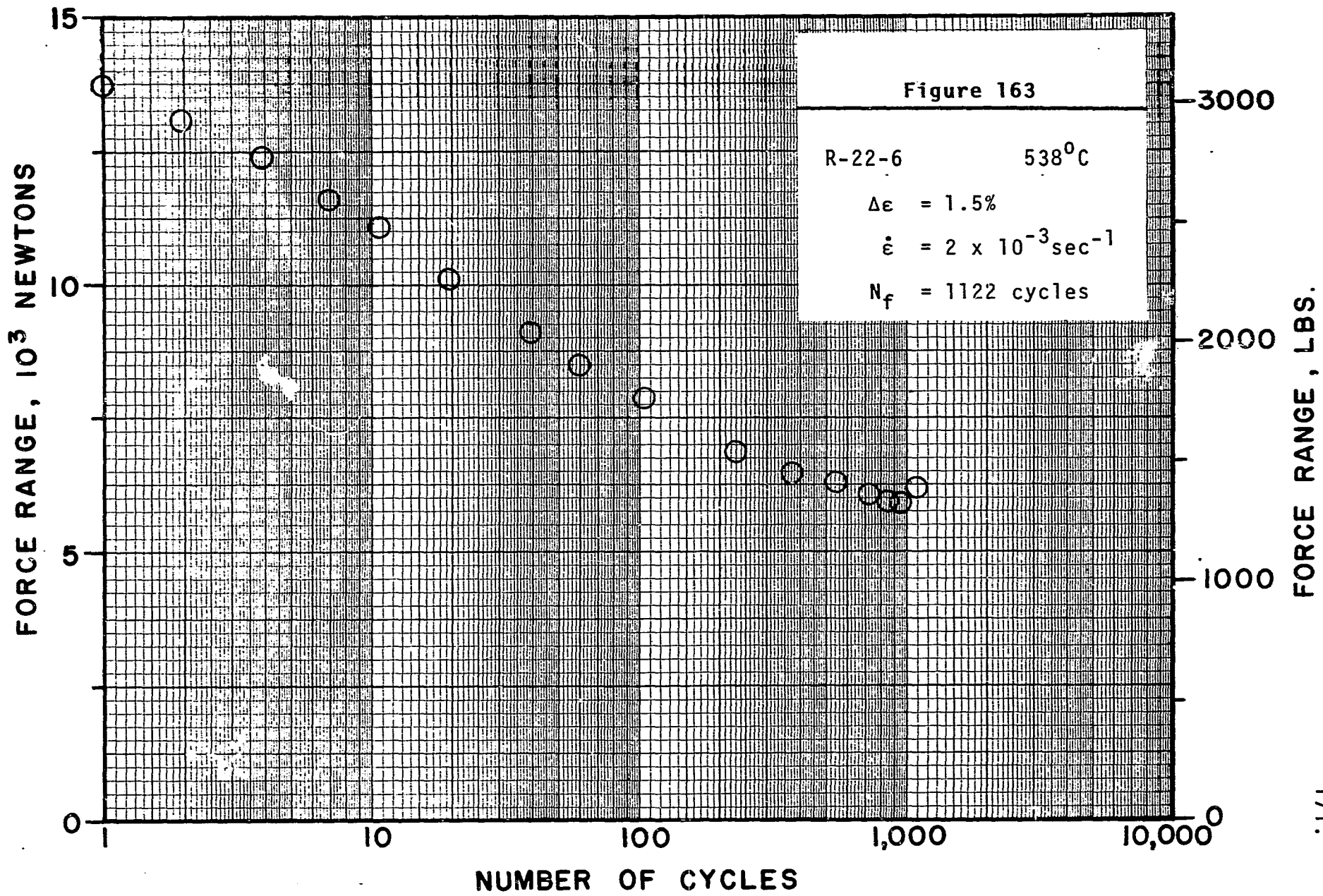


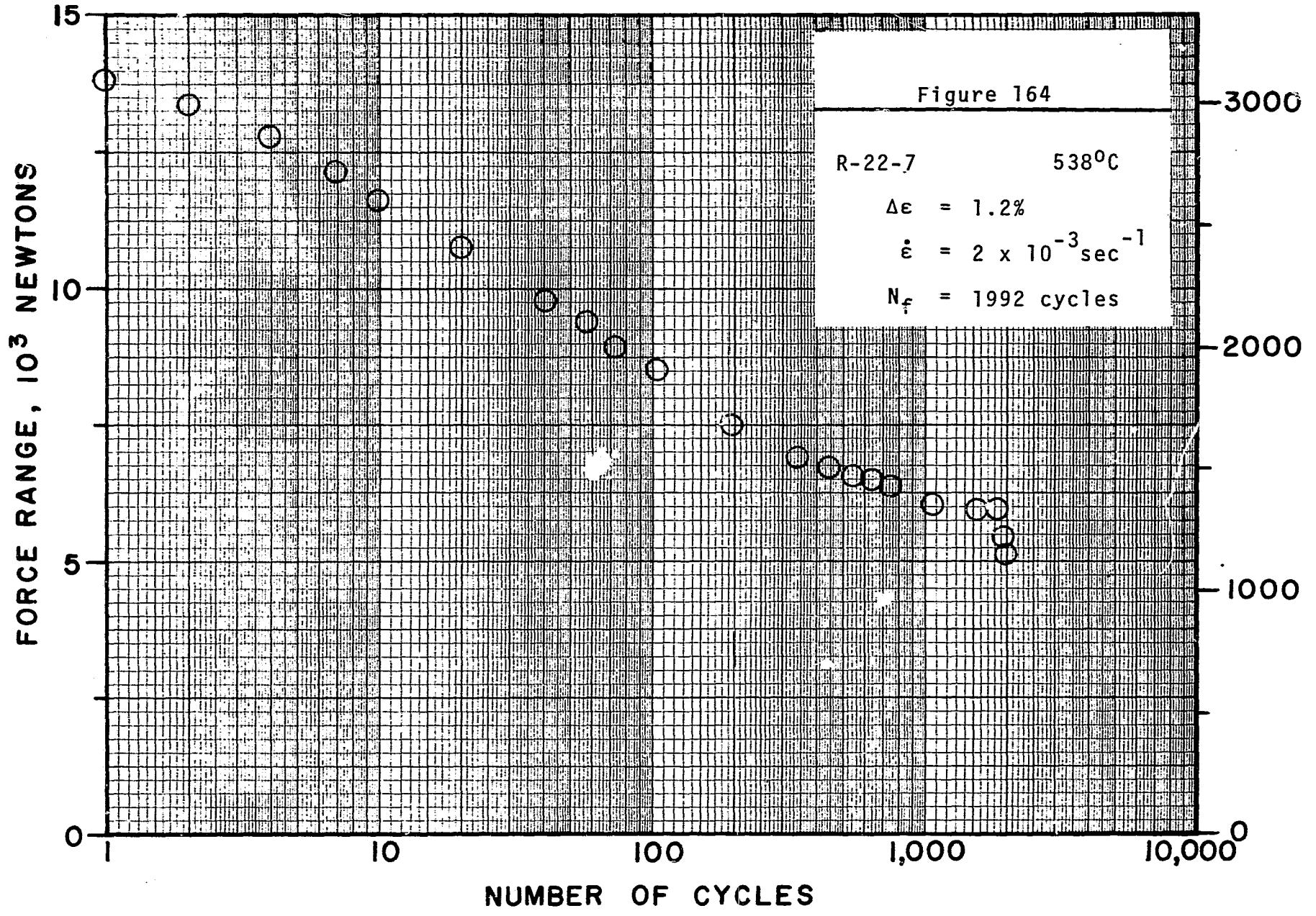








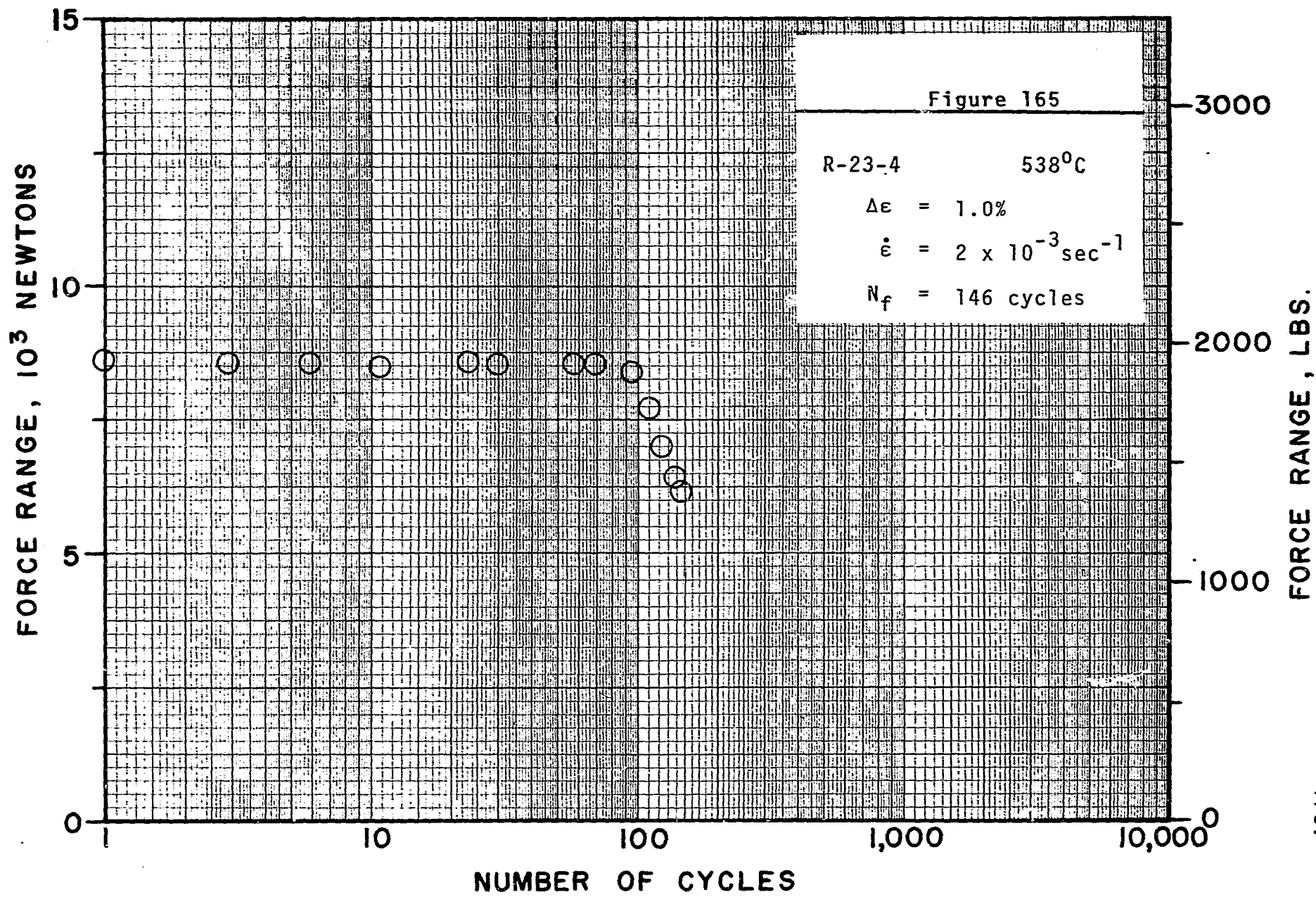


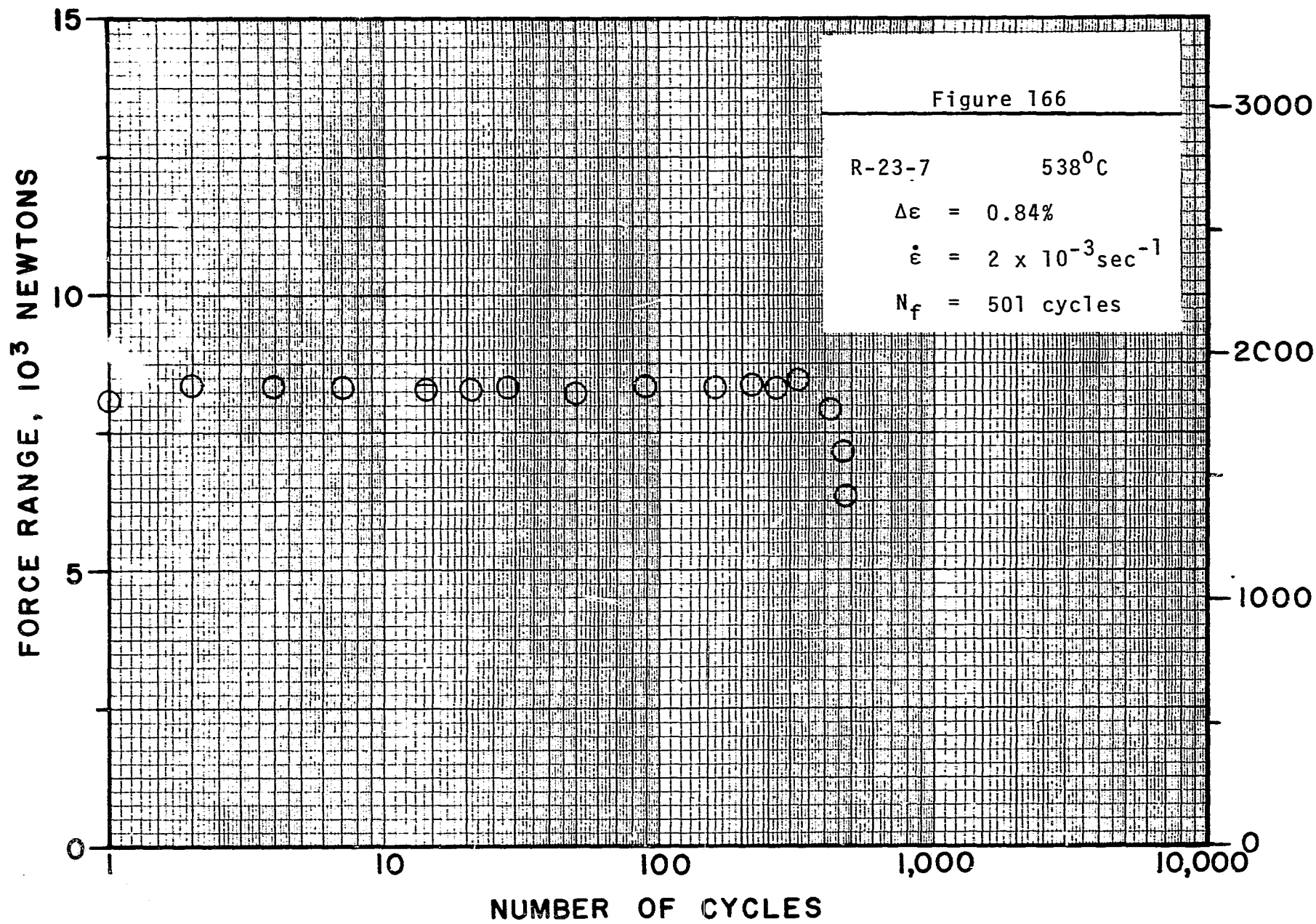


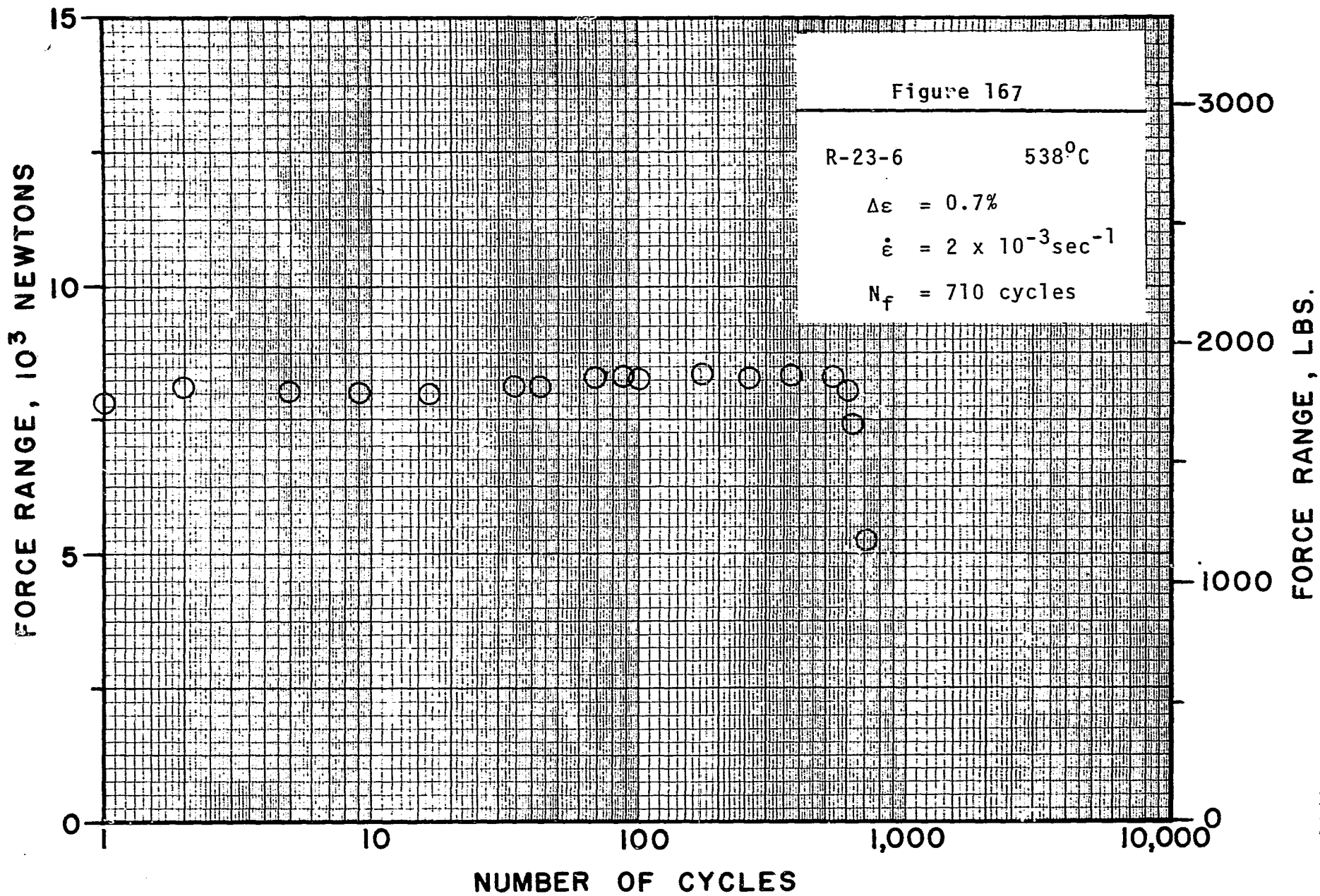
FORCE RANGE, LBS.

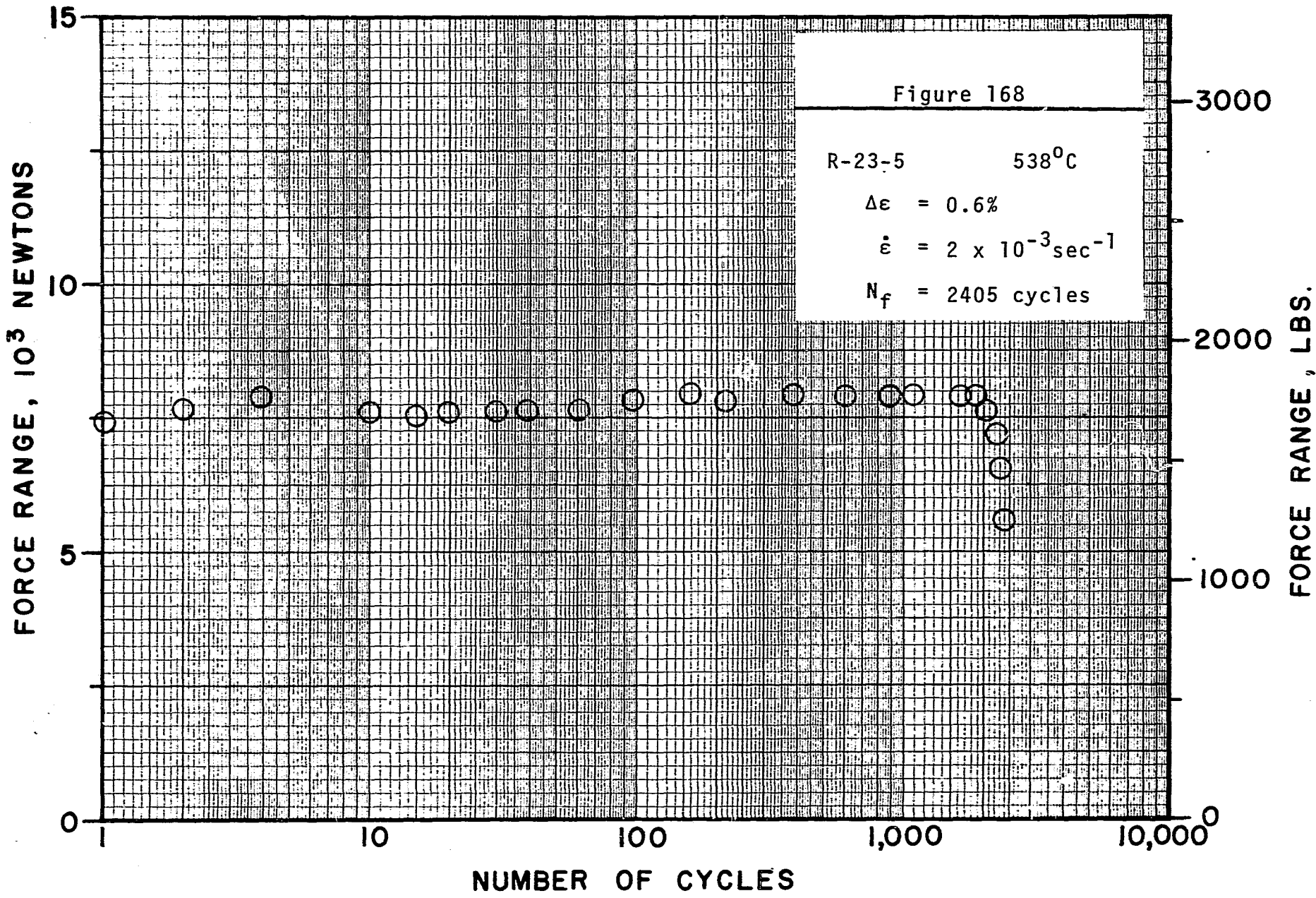
FORCE RANGE, 10³ NEWTONS

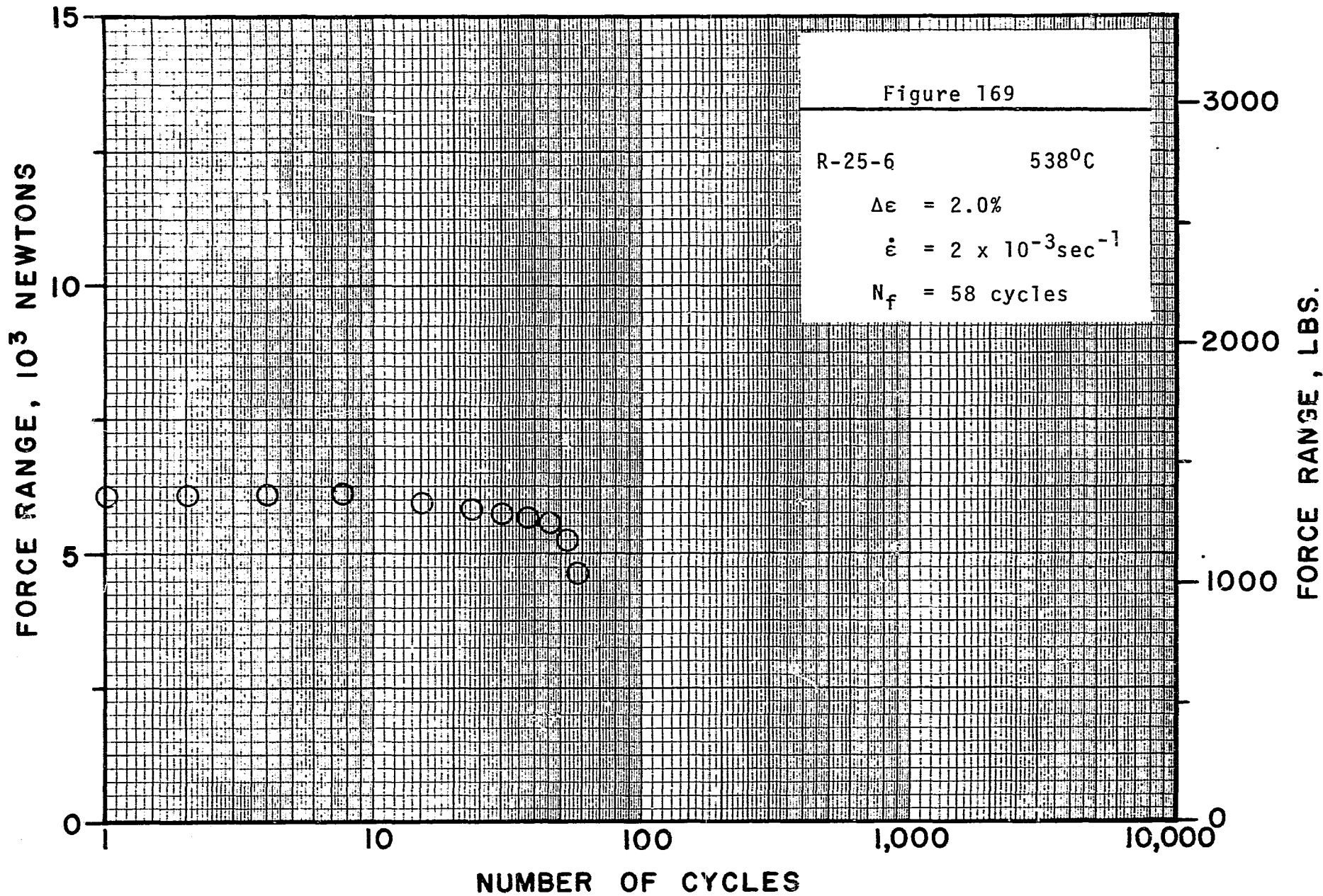
NUMBER OF CYCLES

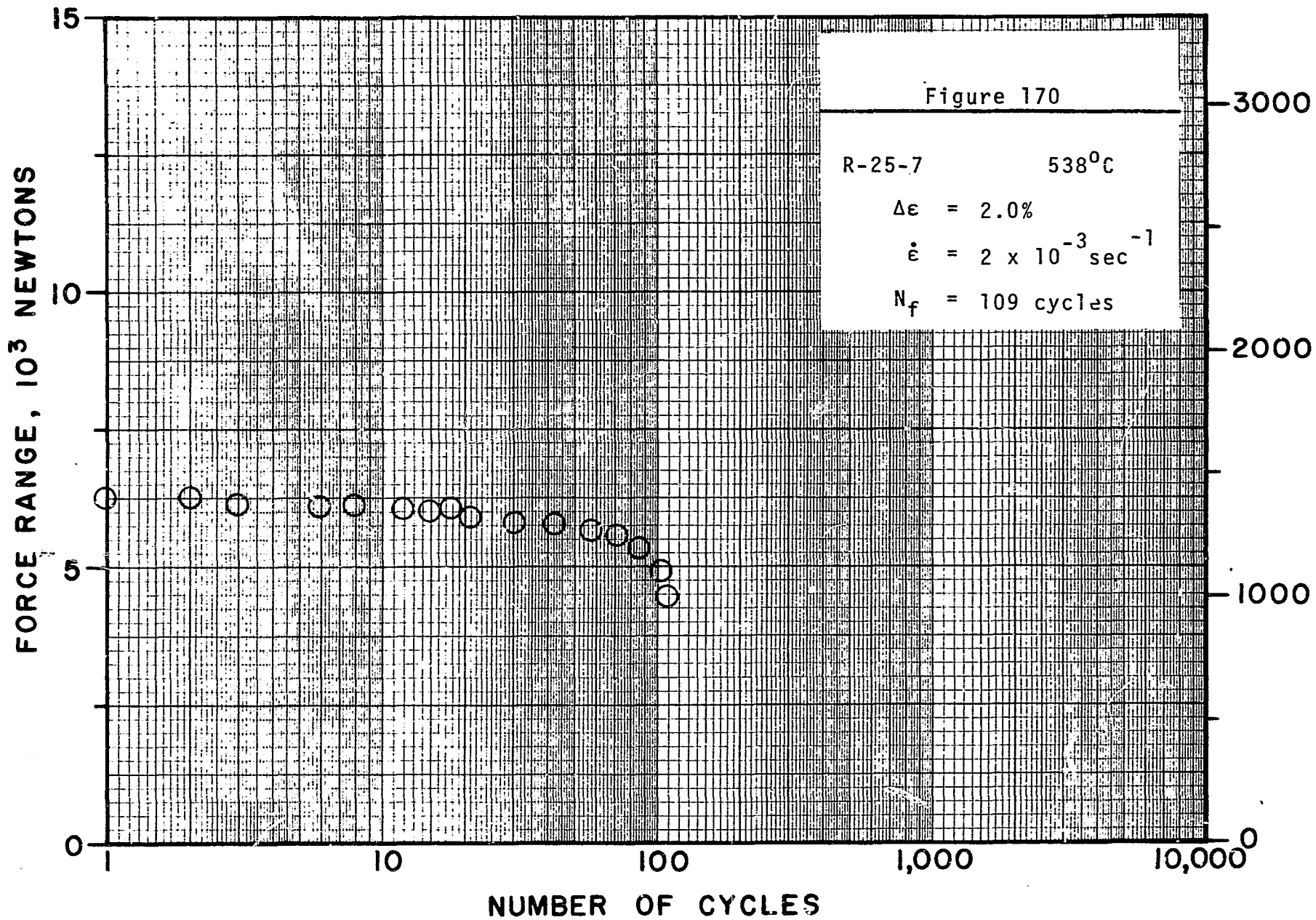


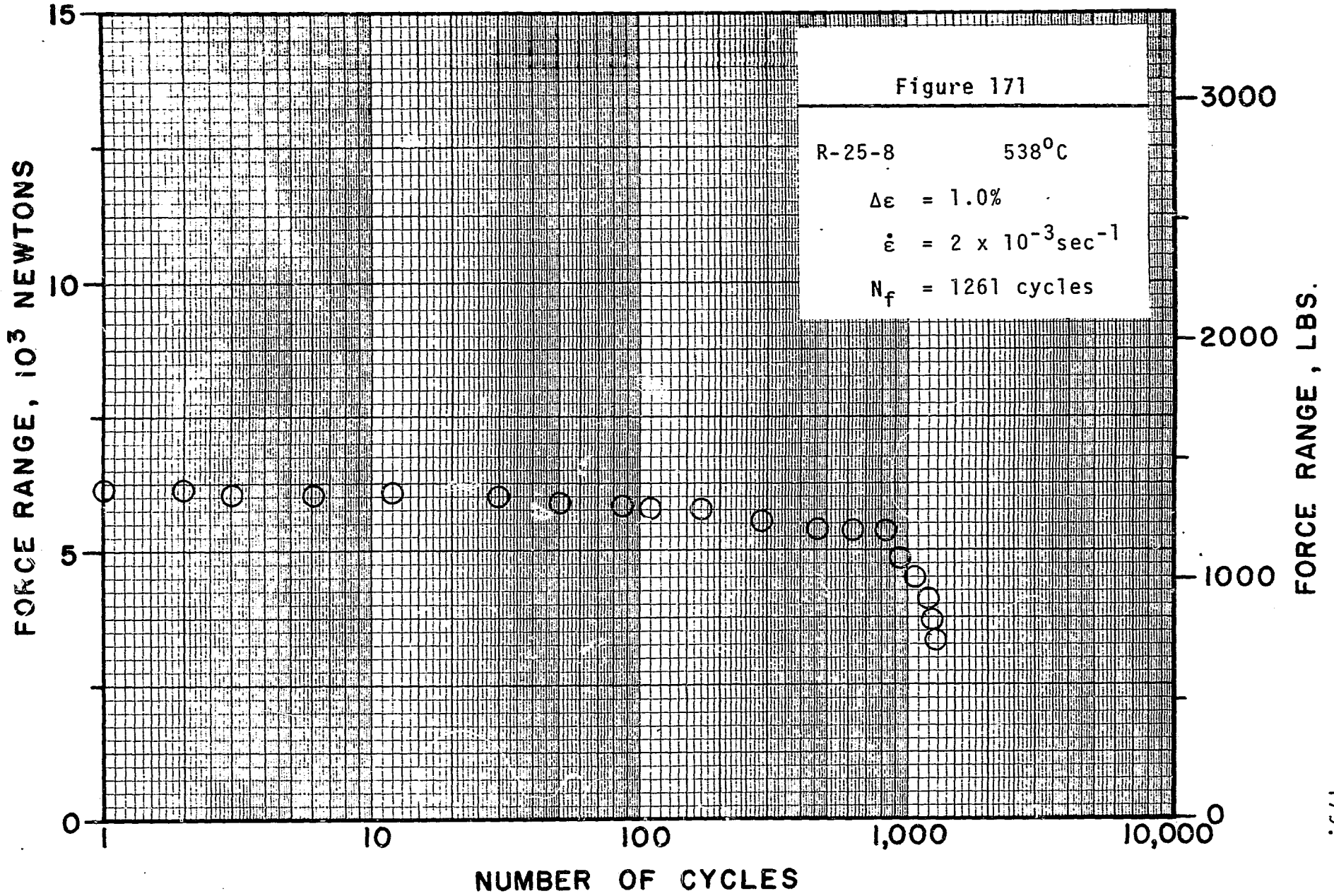


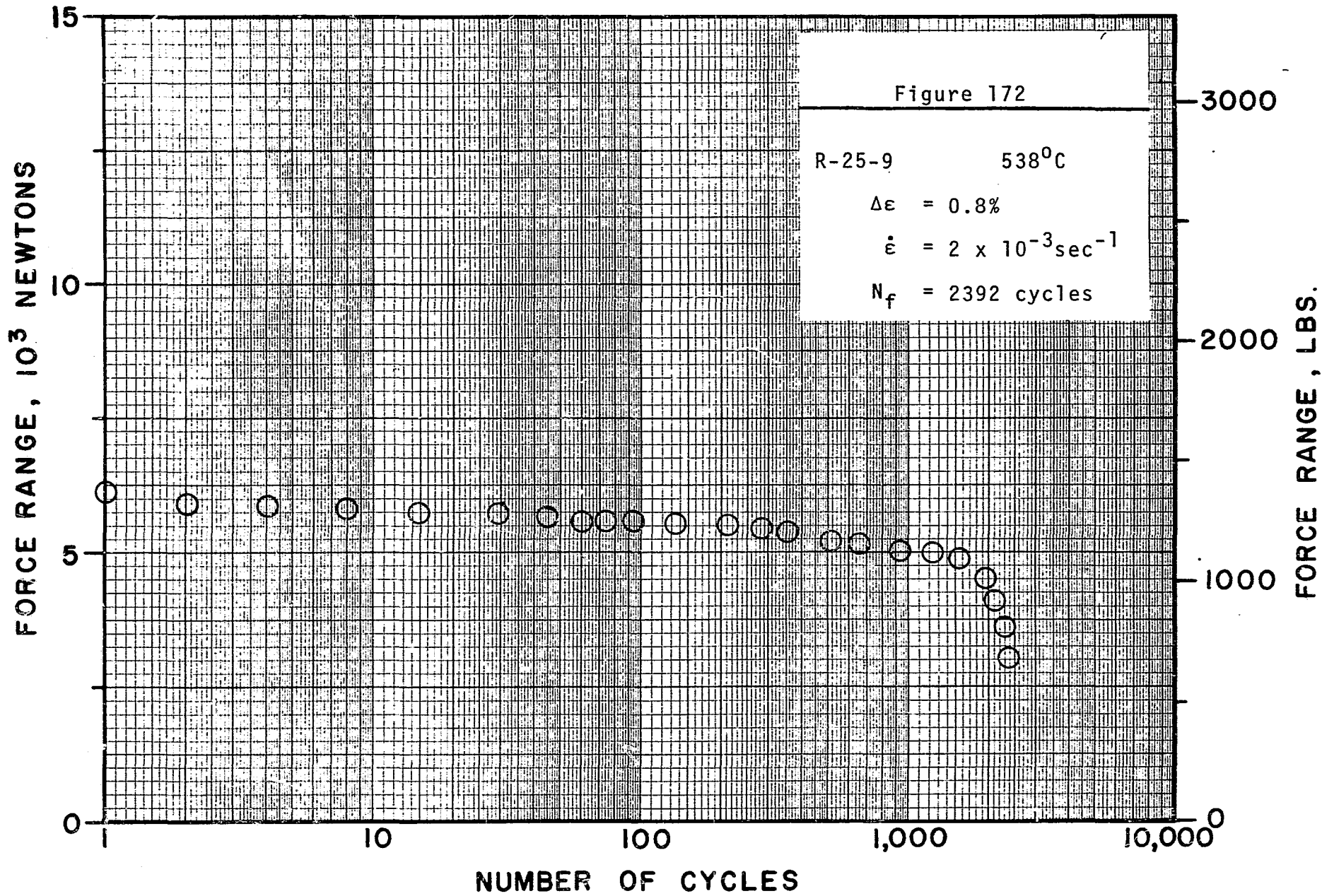


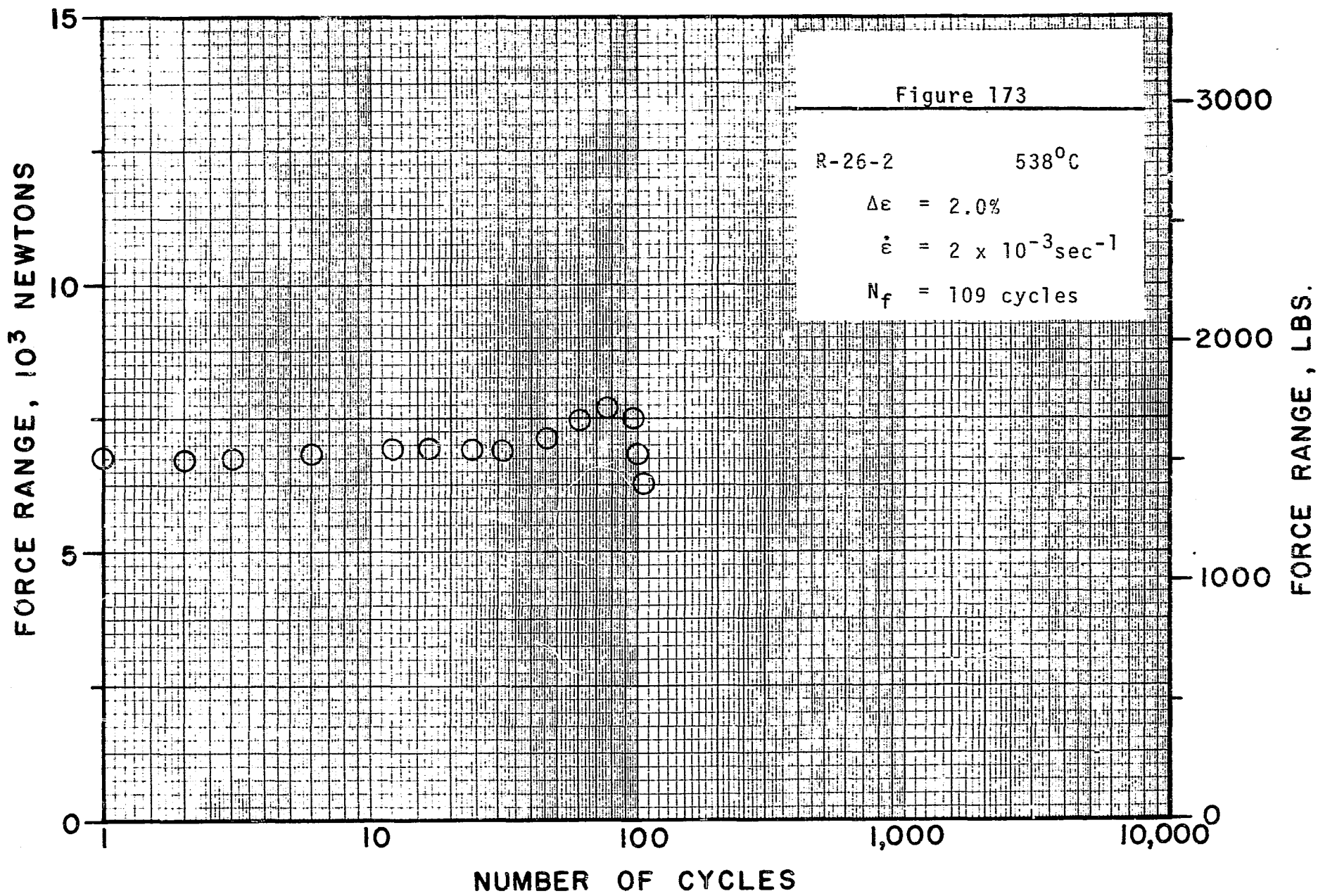


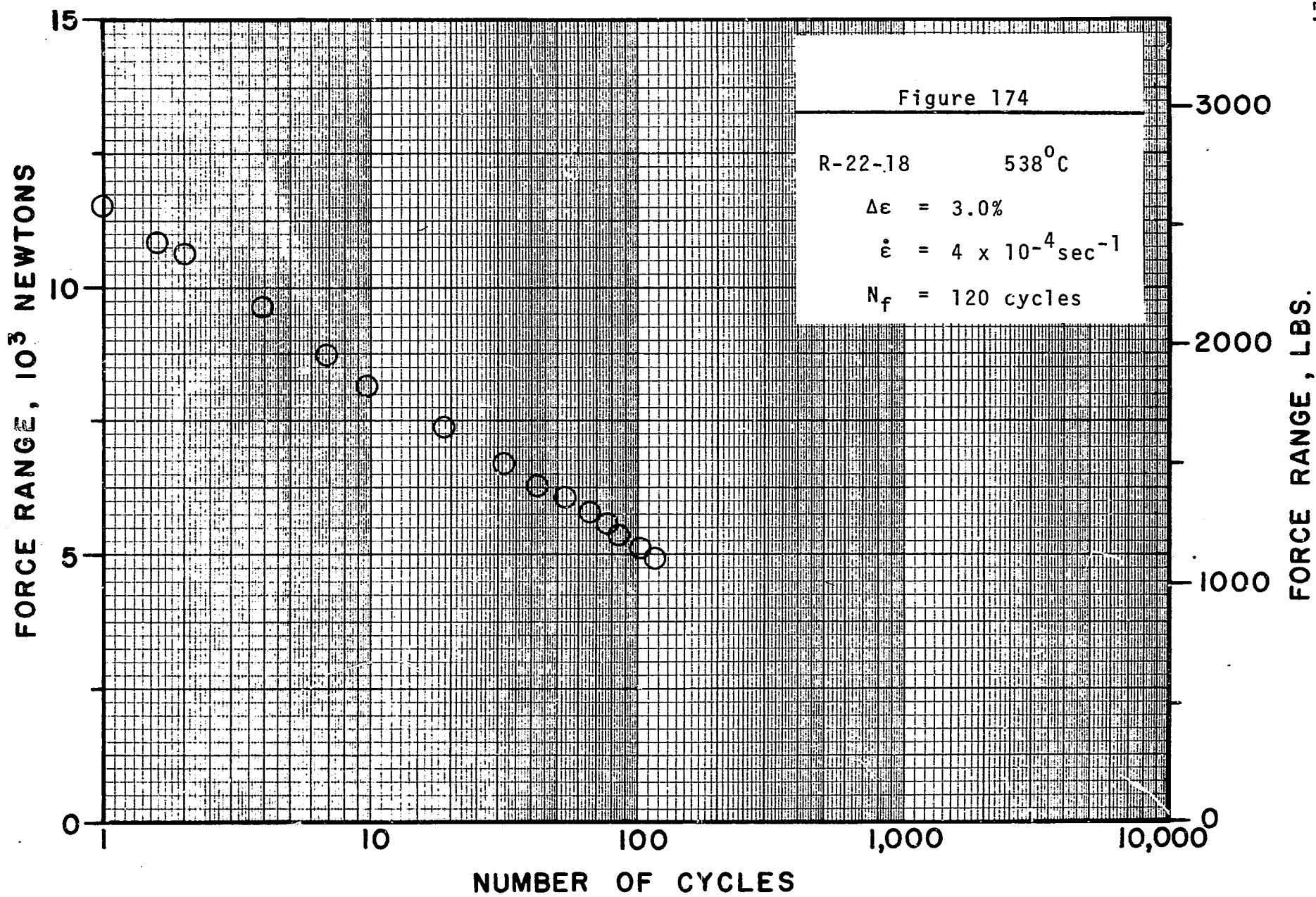


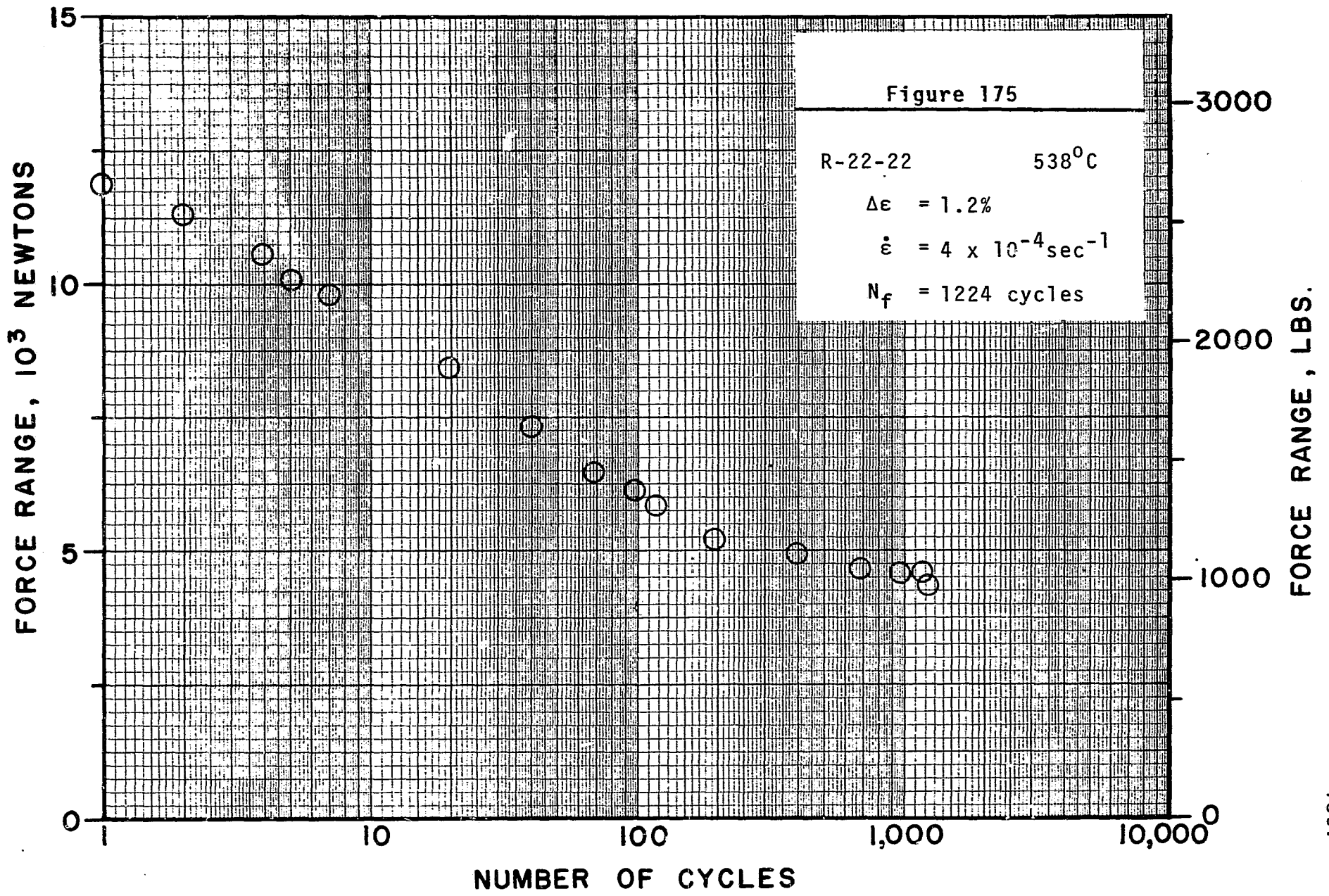


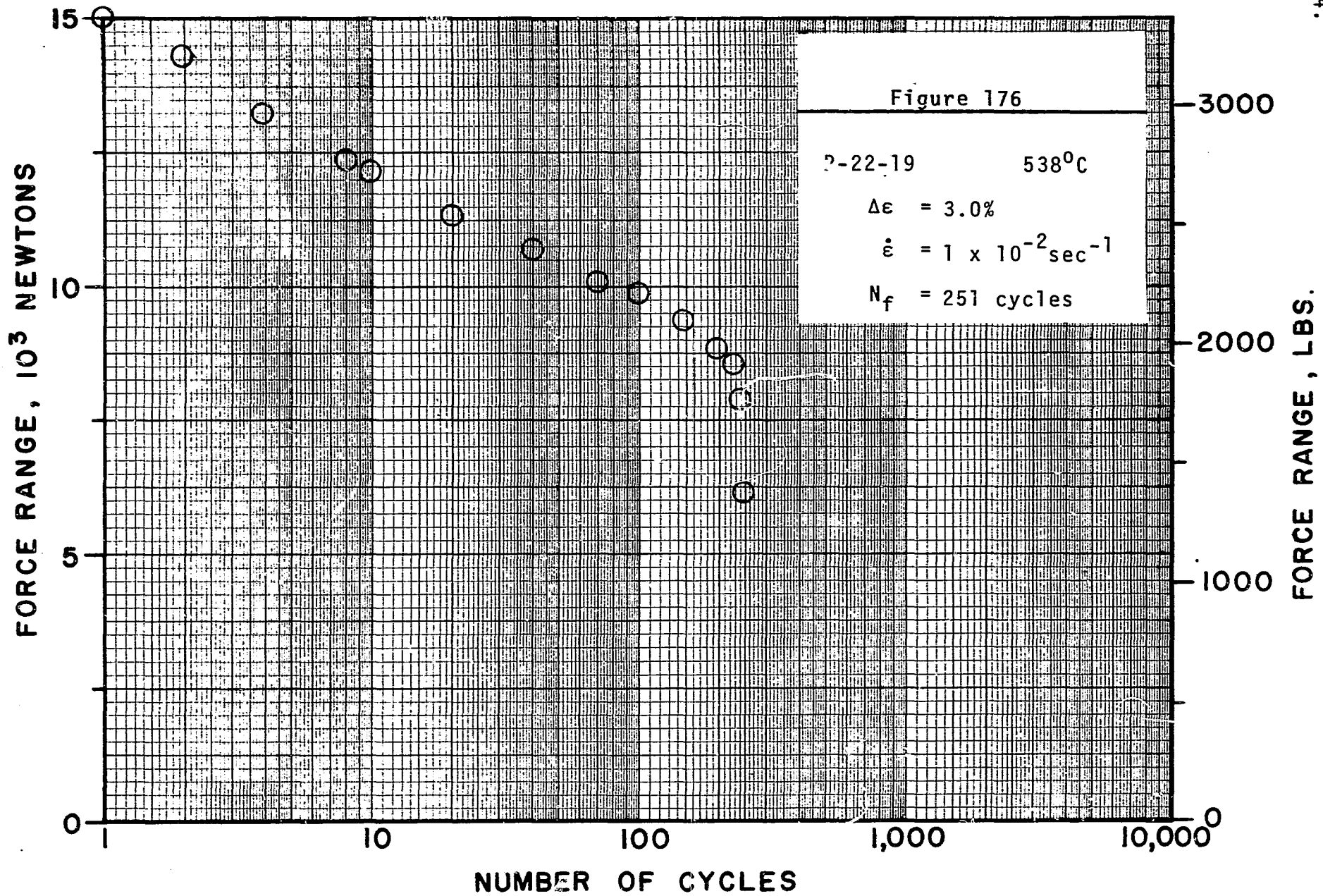


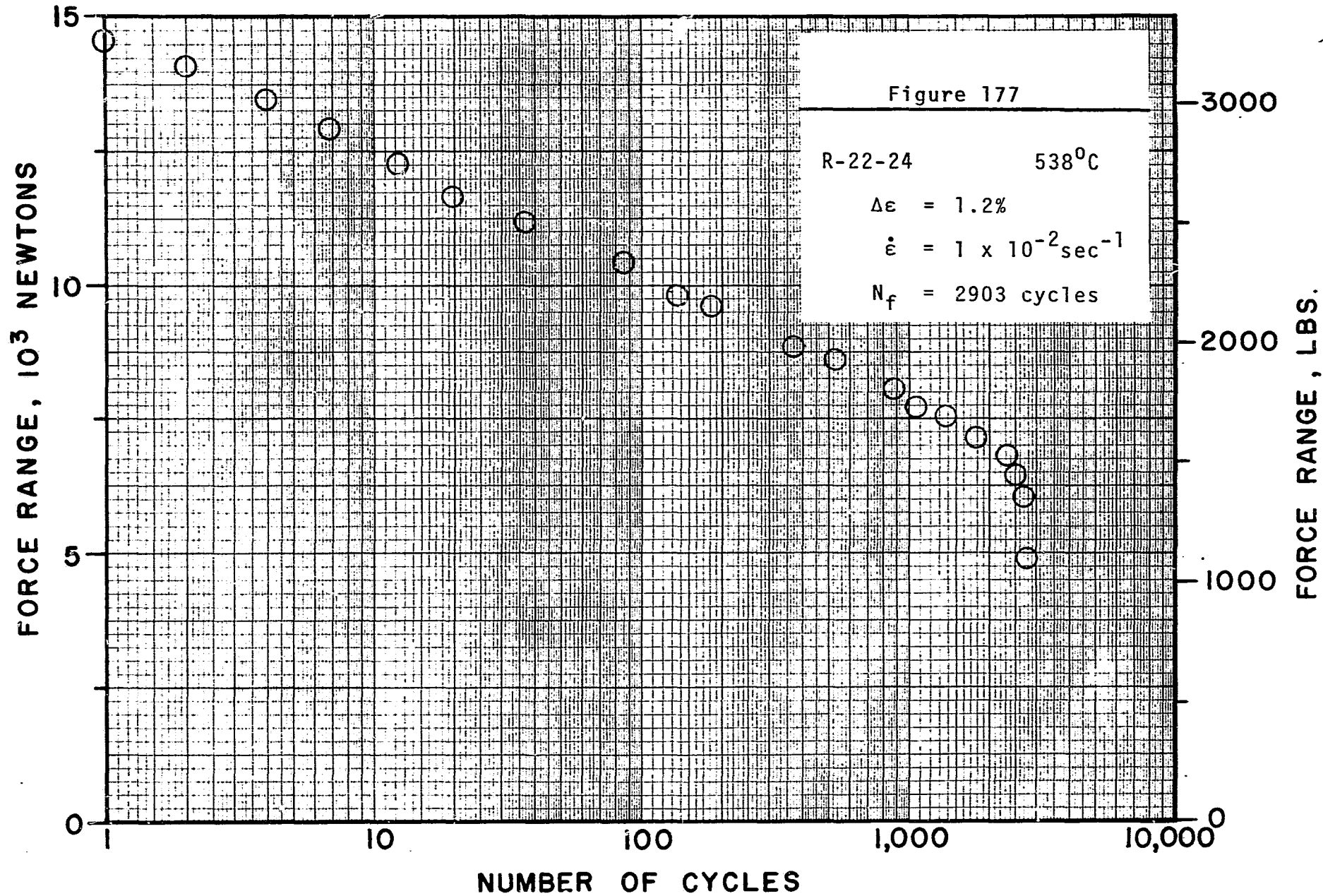


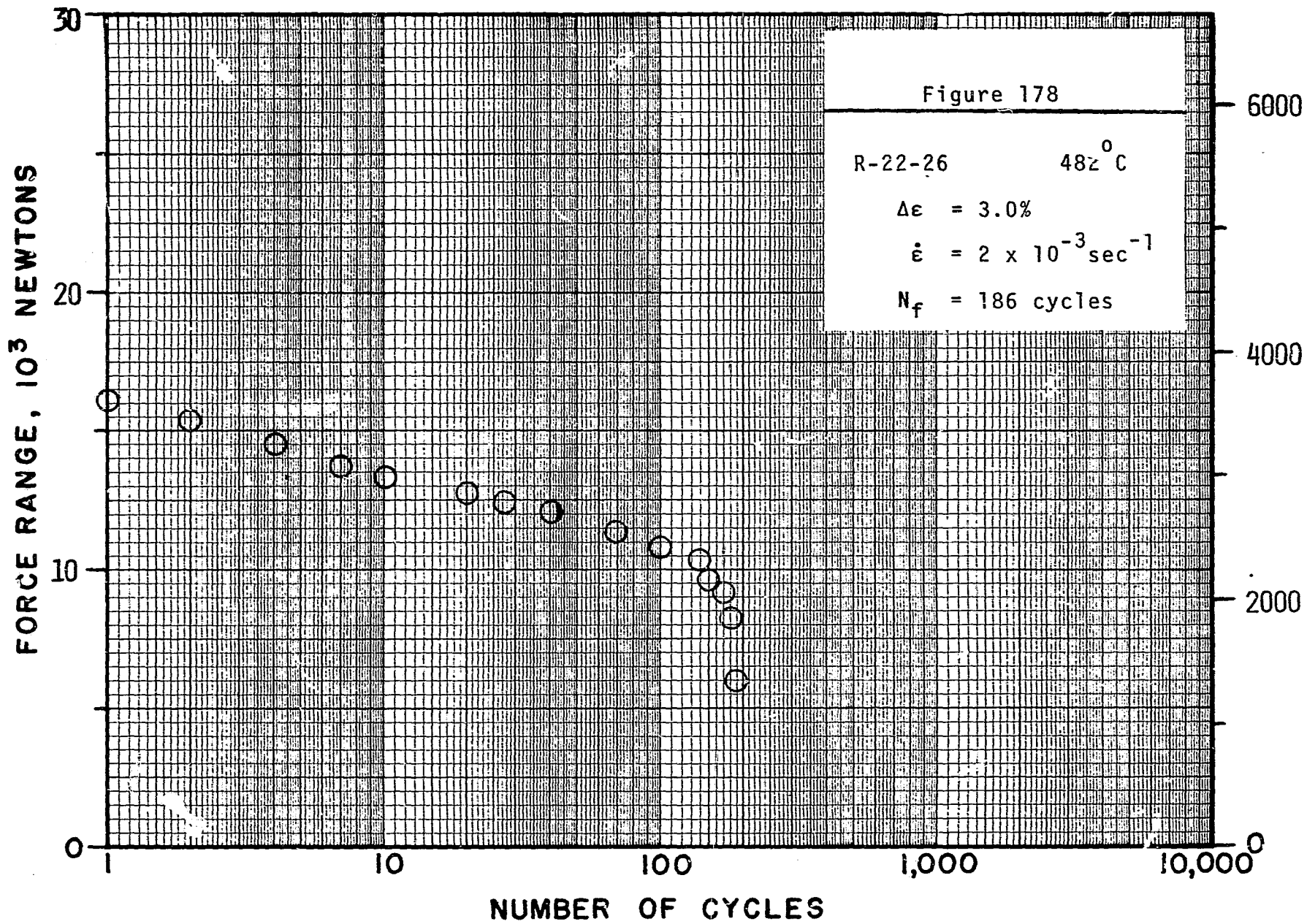




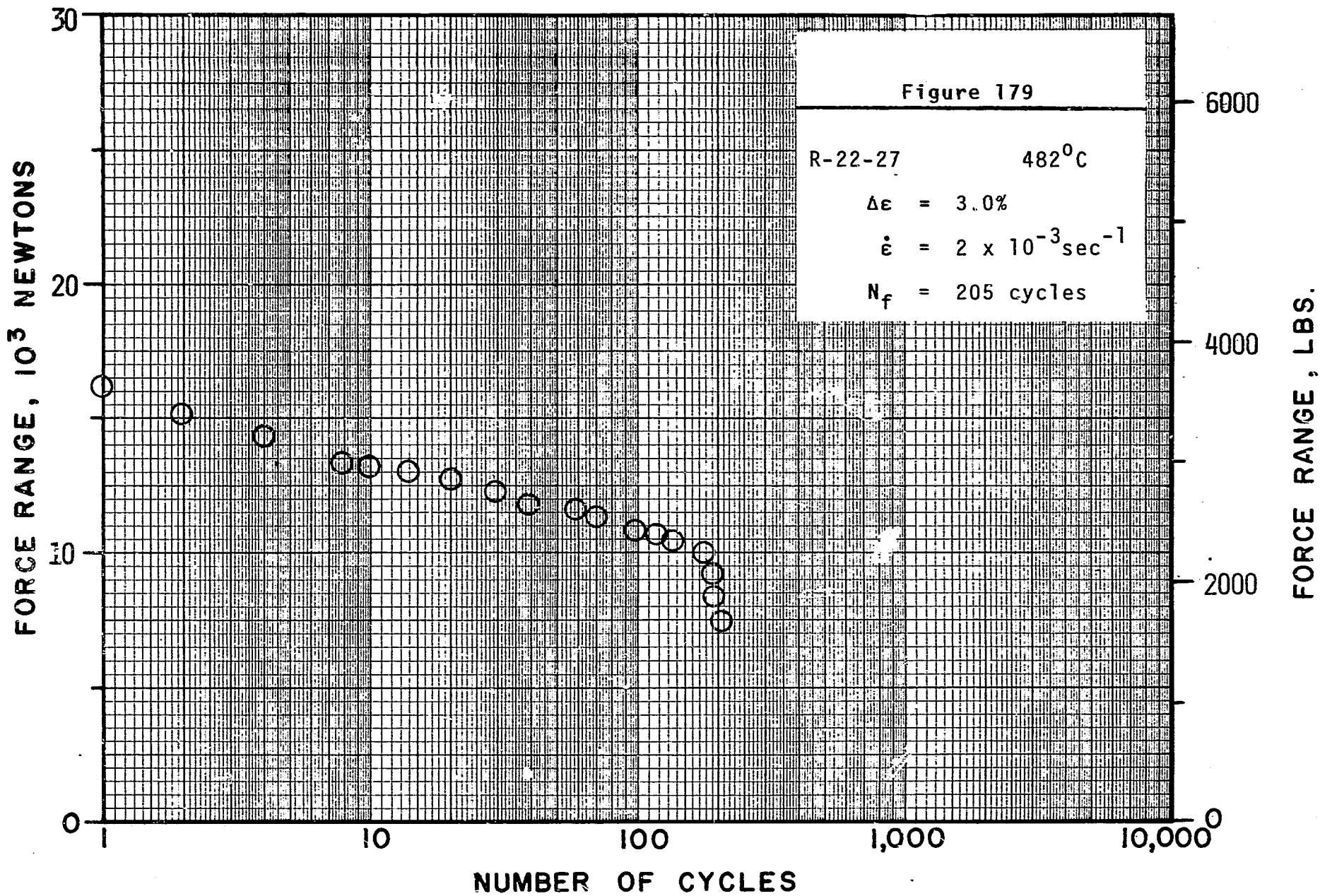


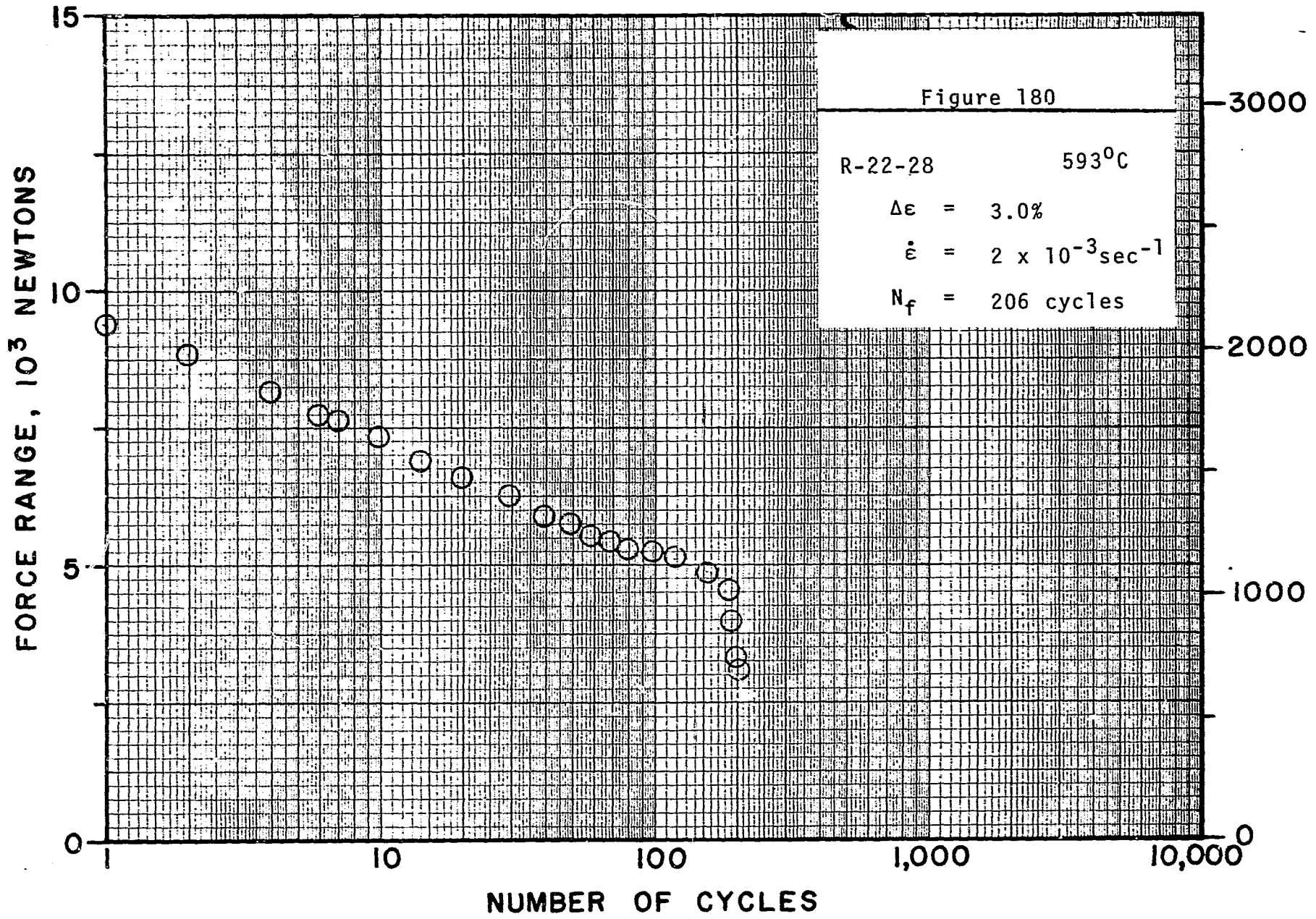


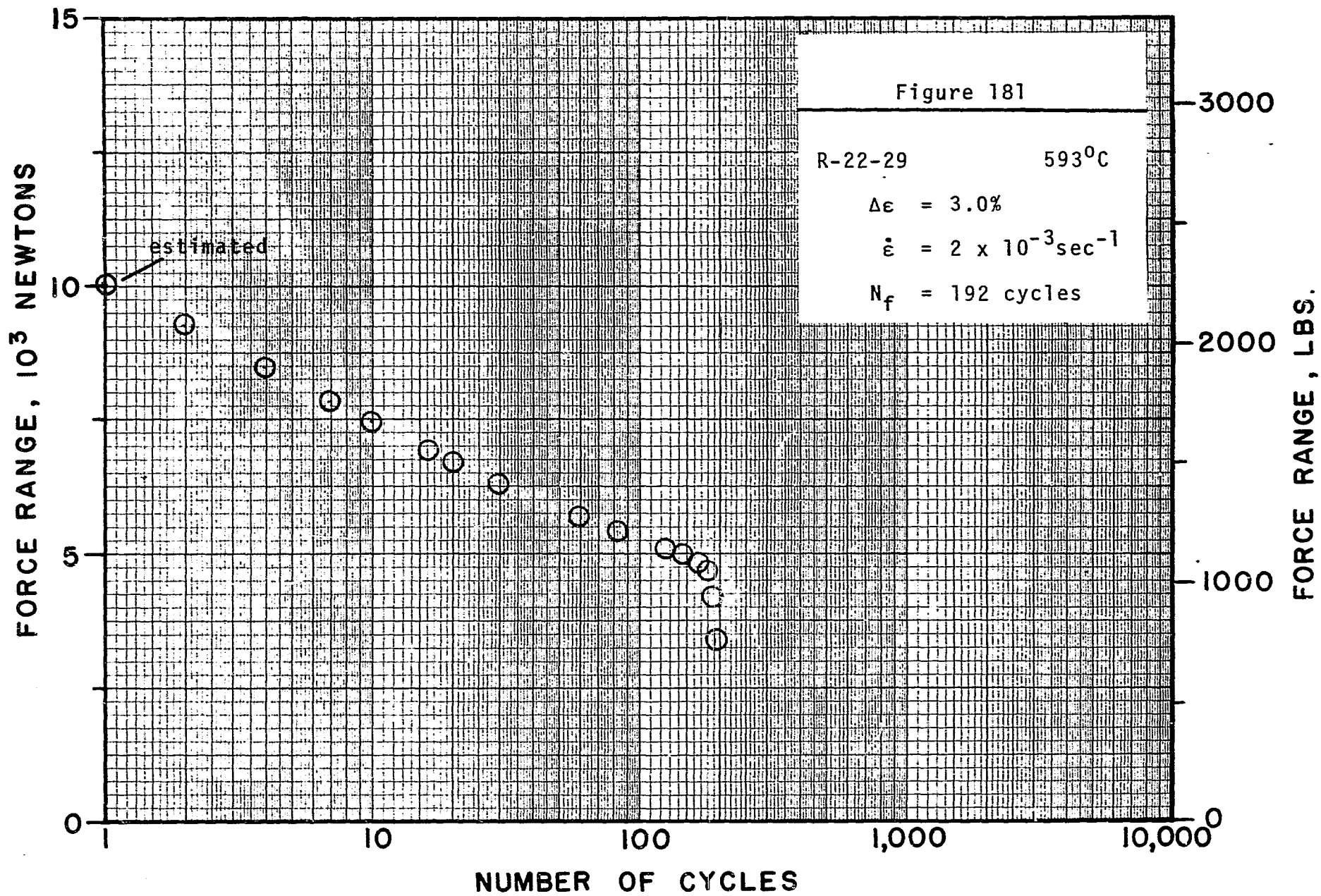


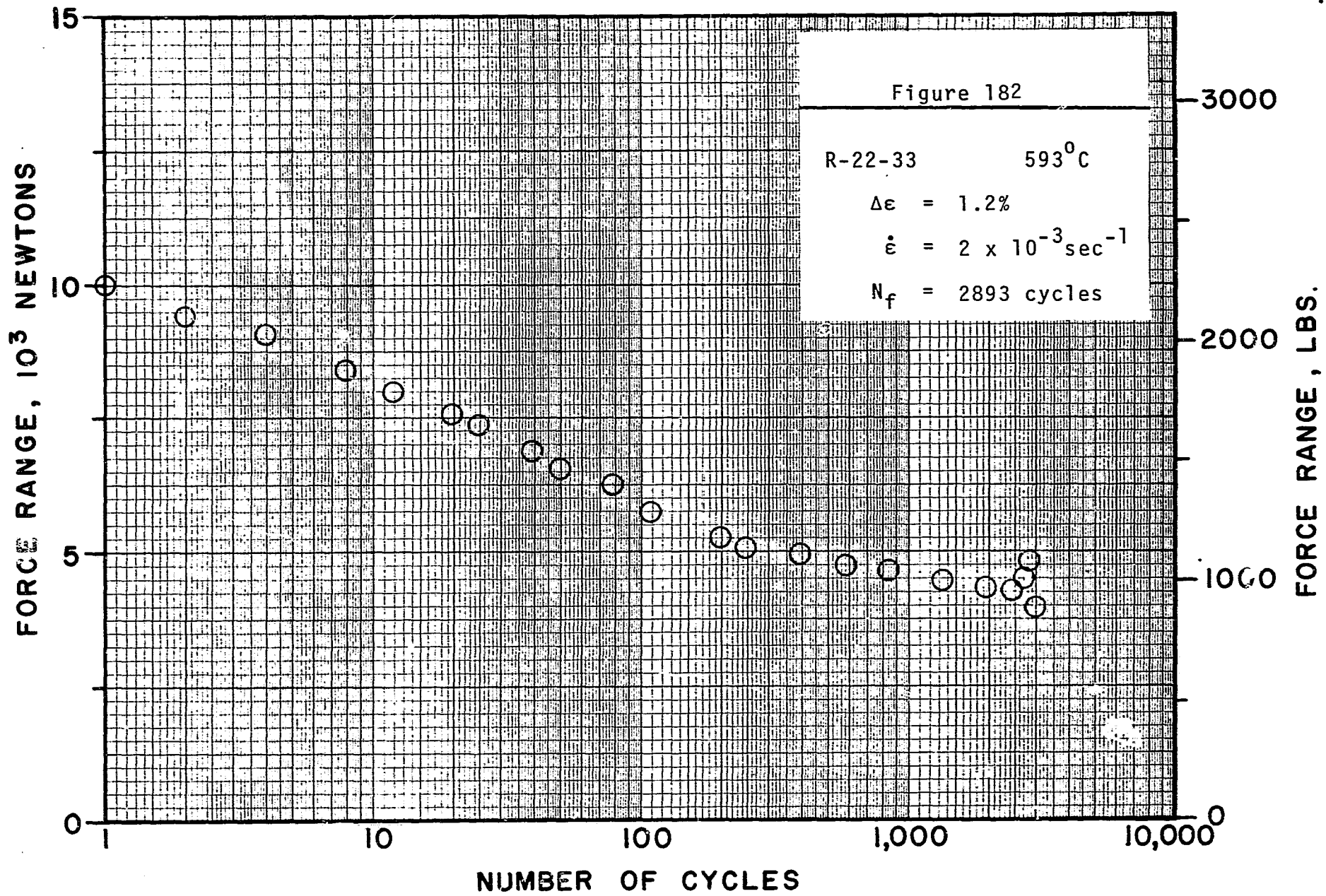


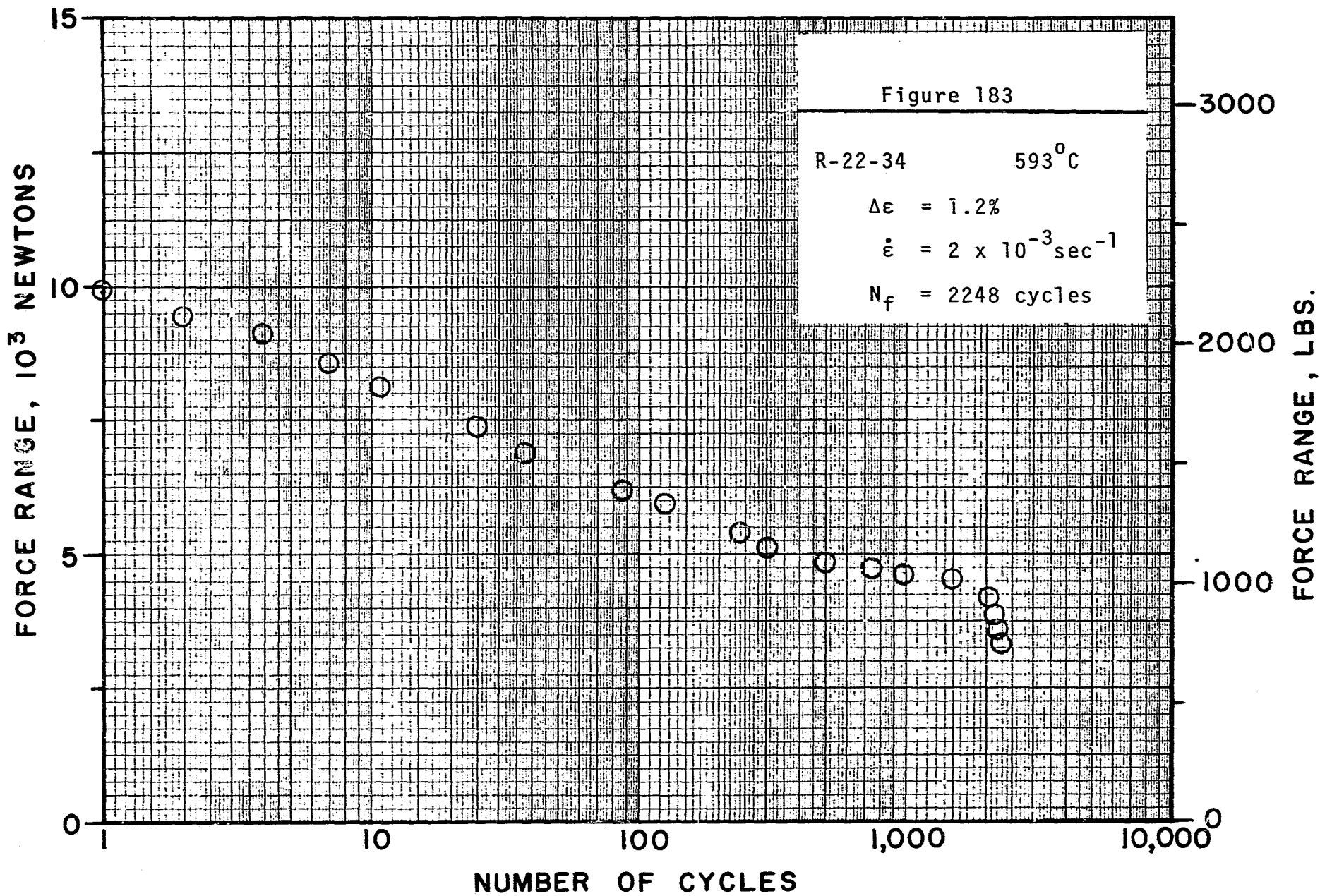
FORCE RANGE, LBS.



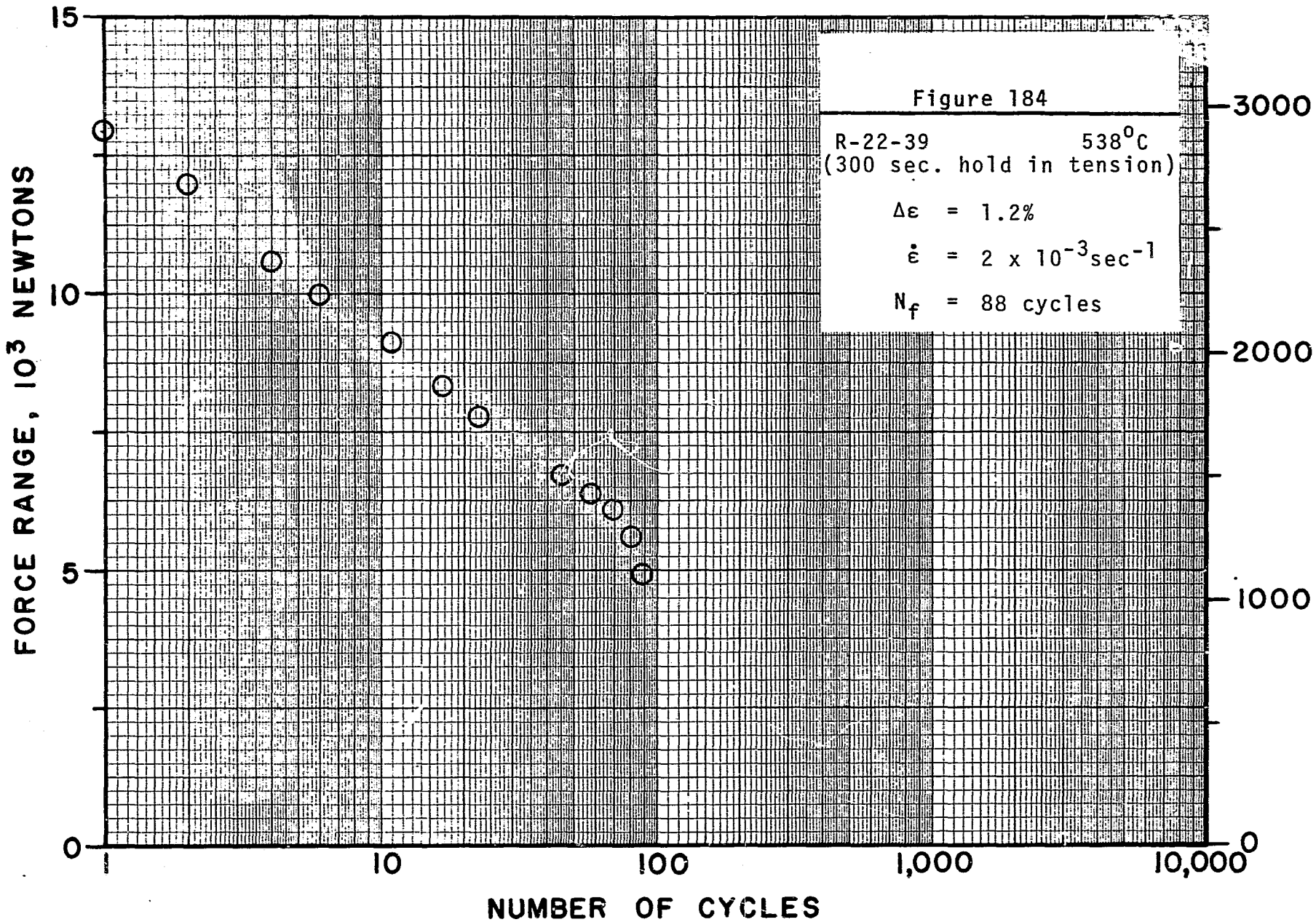


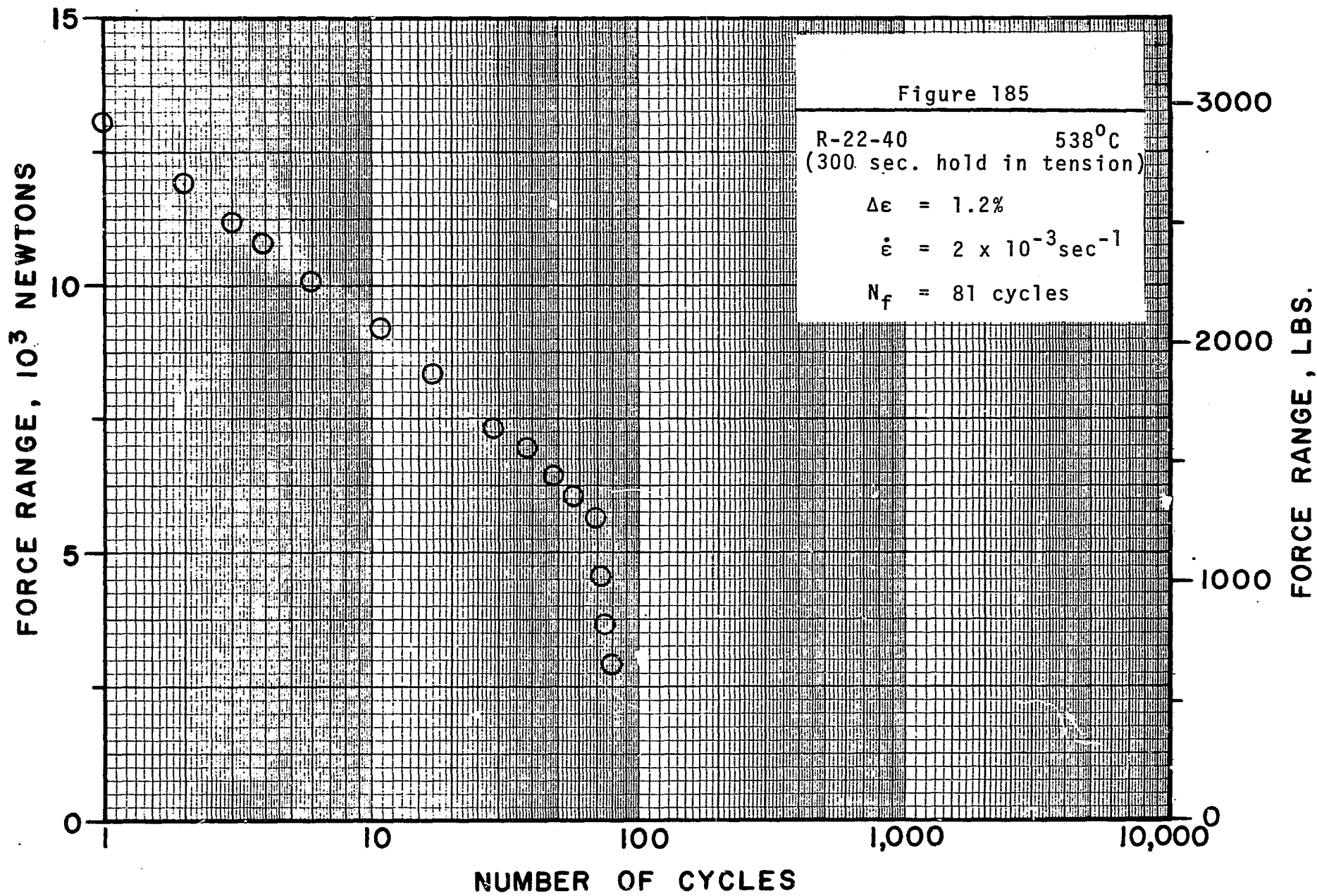


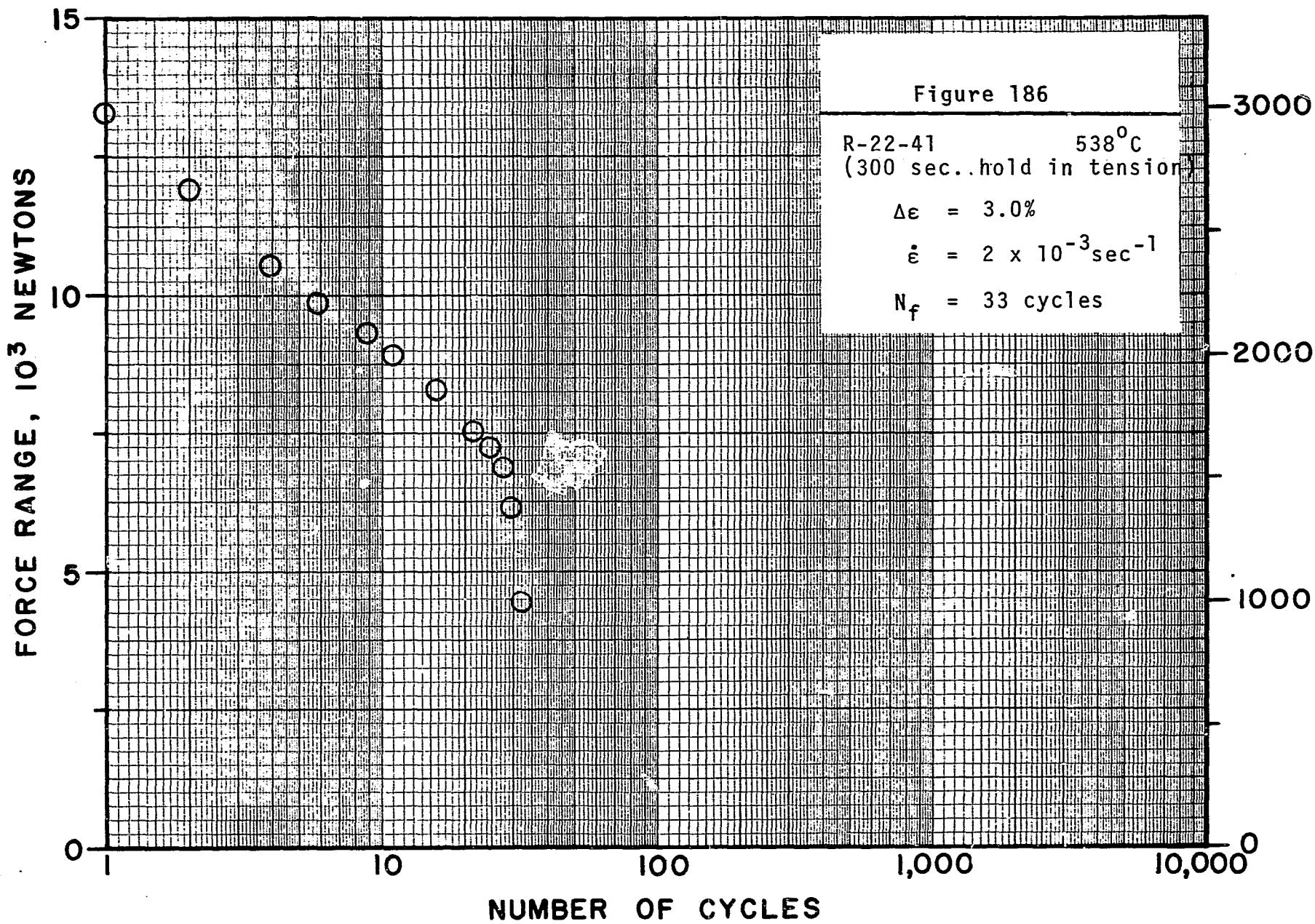


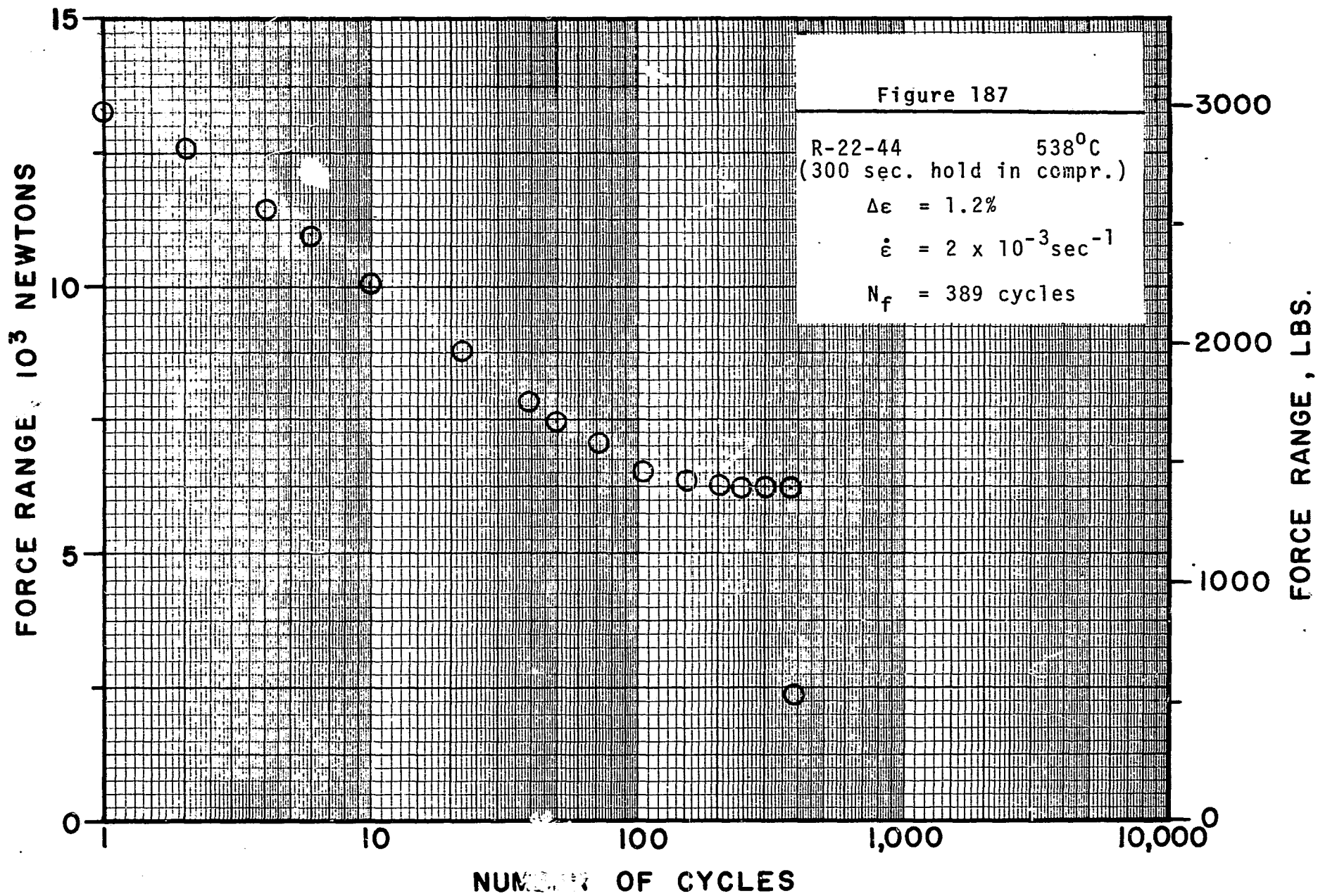


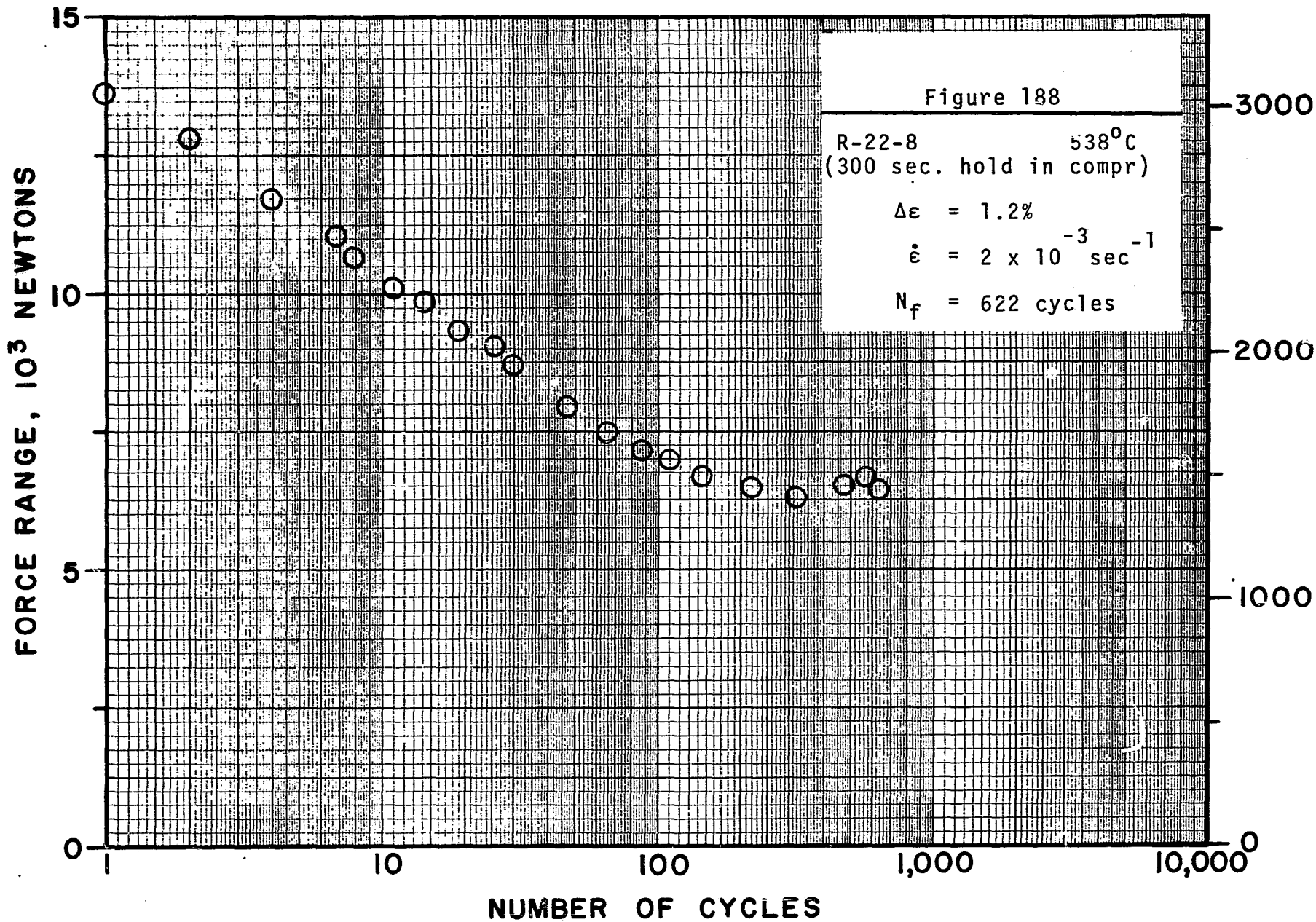
Handwritten initials

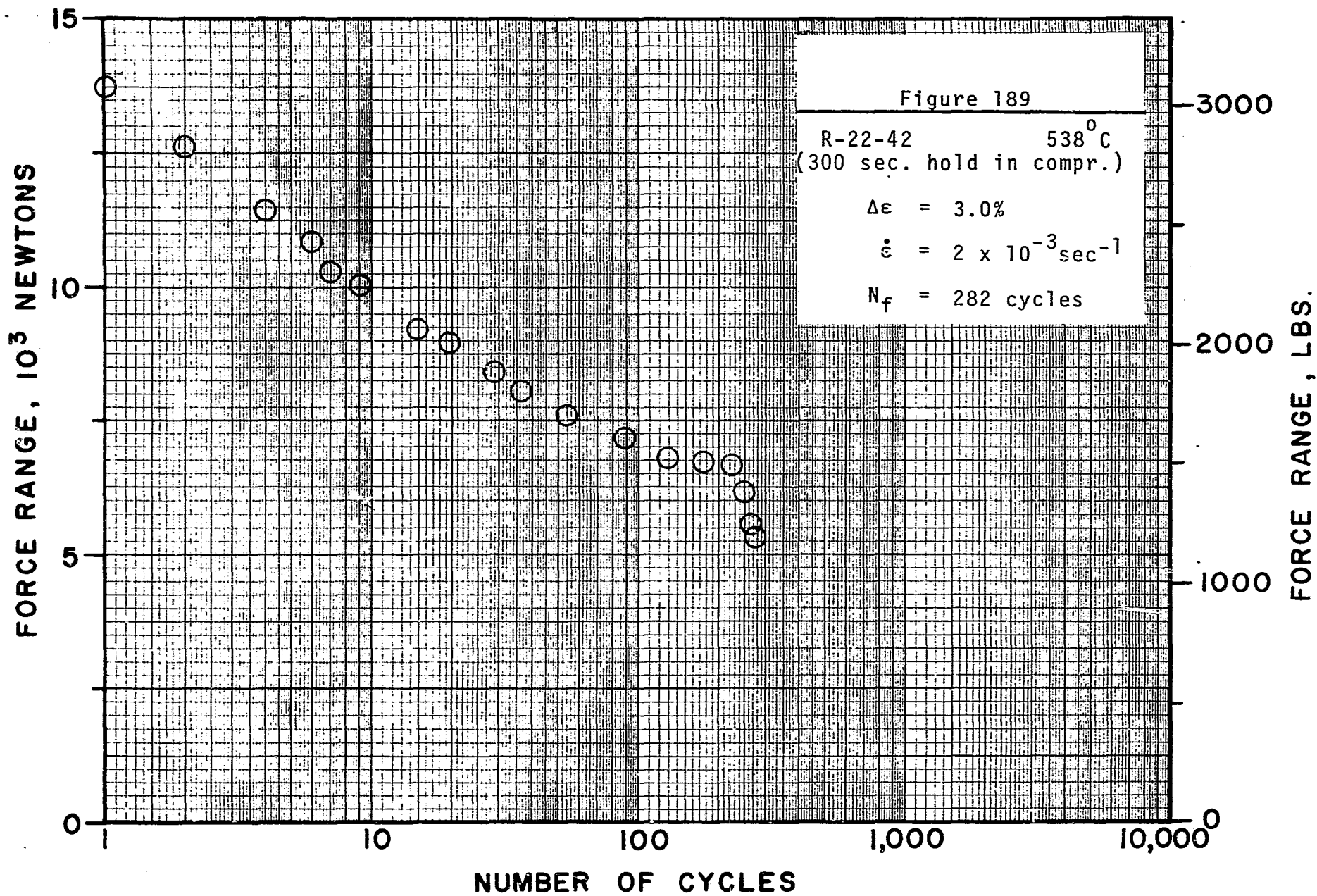












b) HYSTERESIS LOOPS

Figures 190 through 288.

Note: A frequency of 60 cpm was employed in the testing of Specimen Numbers R-24-29, R-24-30, R-2-74 and R-2-75. Since this frequency was beyond the response characteristics of the x-y recorder, no hysteresis loops could be obtained for these tests.

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

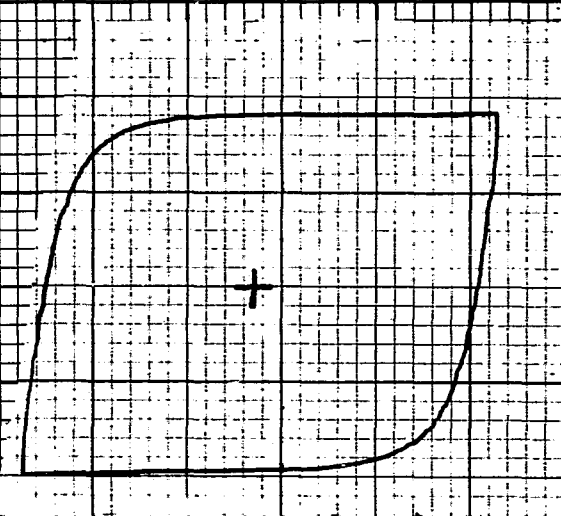
TENSION

LBS
NEWTONS

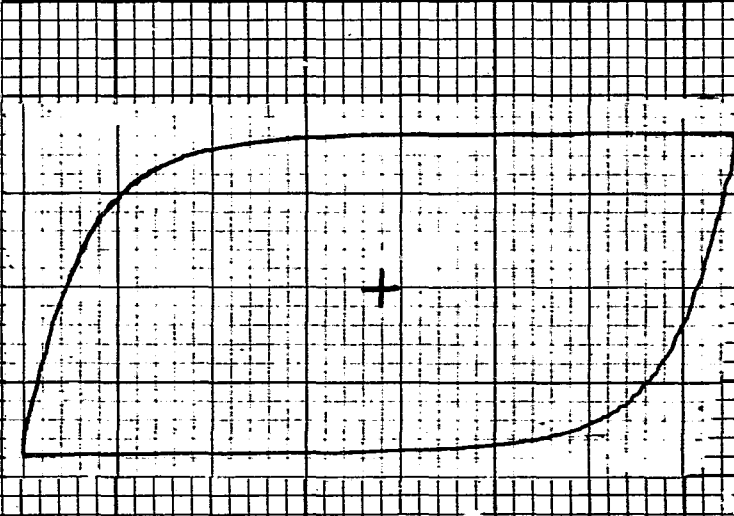
1000
500
0
500
1000

4000
2000
0
2000
4000

COMPRESSION



R-0-14 538°C
 Cycle Number: 102
 N_f : 1512 cycles
 $\Delta\epsilon$: 2.0%



R-0-15 538°C
 Cycle Number: 600
 N_f : 4188 cycles
 $\Delta\epsilon$: 1.5%

Figure 190

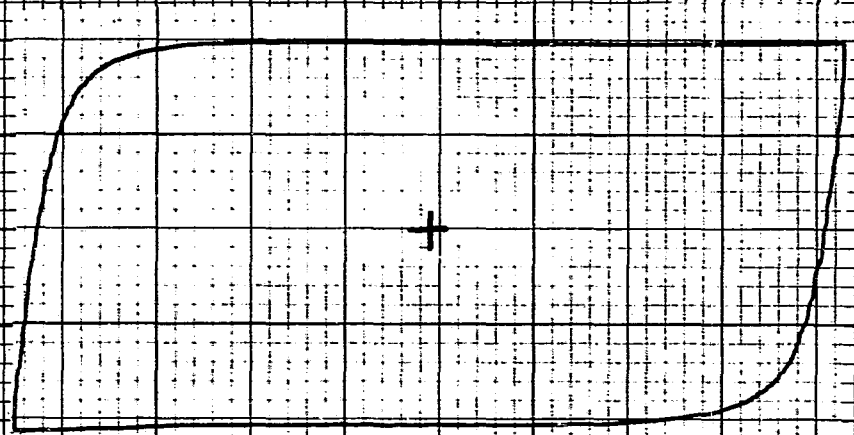
HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION

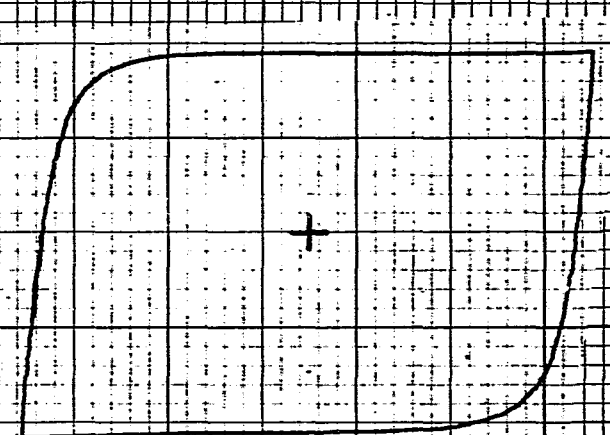
1000
 500
 0
 500
 1000

4000
 2000
 0
 2000
 4000

LBS
 NEWTONS



R-0-25 538°C
 Cycle Number: 58
 N_f : 283 cycles
 ΔE : 3.5%



R-0-17 538°C
 Cycle Number: 120
 N_f : 307 cycles
 ΔE : 3.0%

COMPRESSION

Figure 191

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

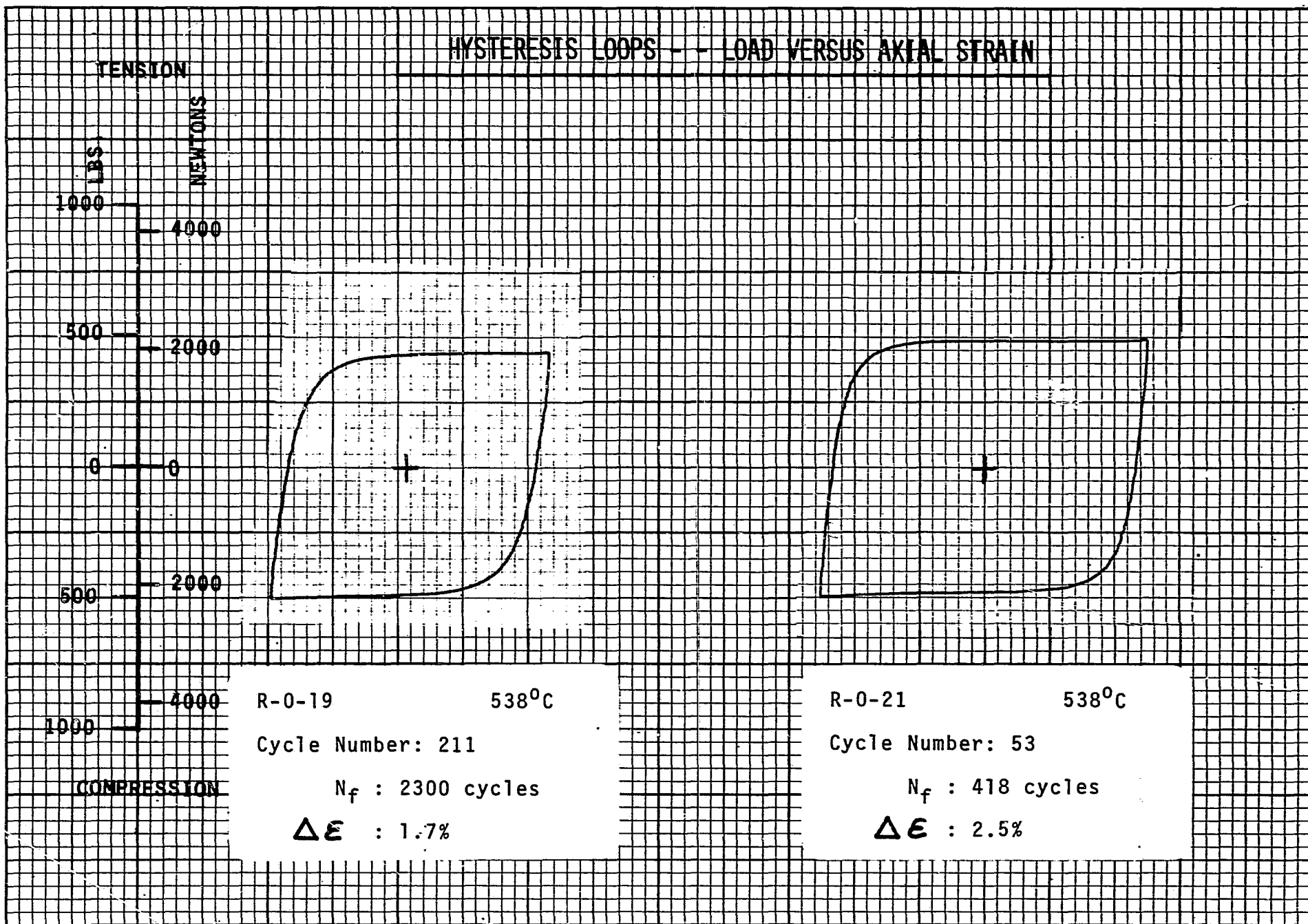


Figure 192

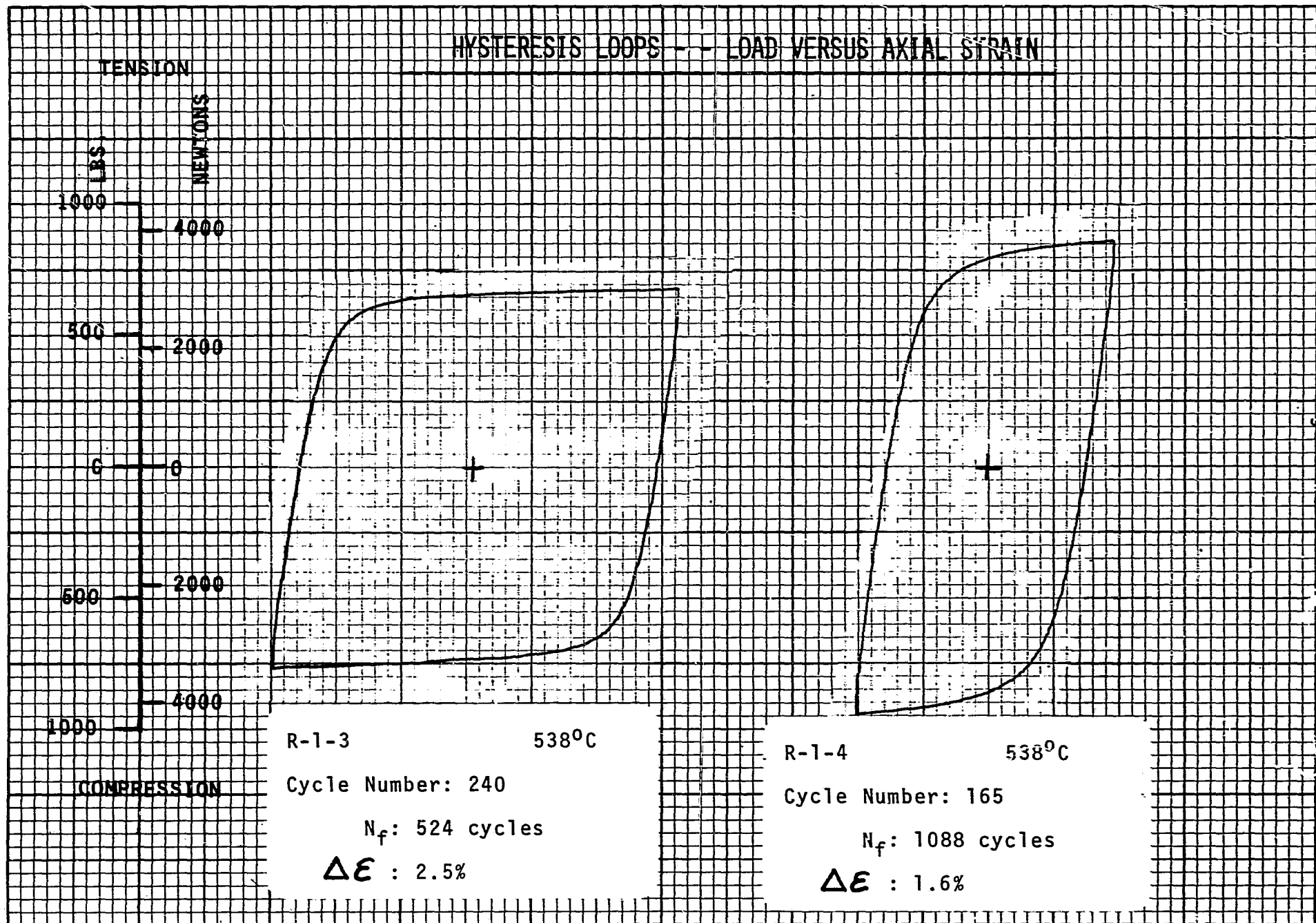


Figure 193

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

LBS

NEWTONS

1000

500

0

500

1000

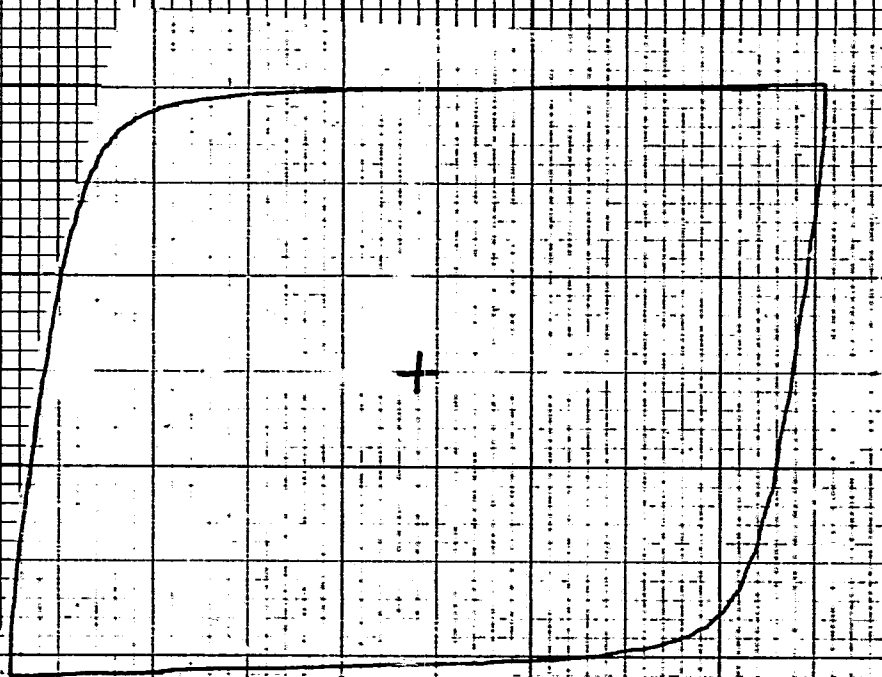
4000

2000

0

2000

4000



COMPRESSION

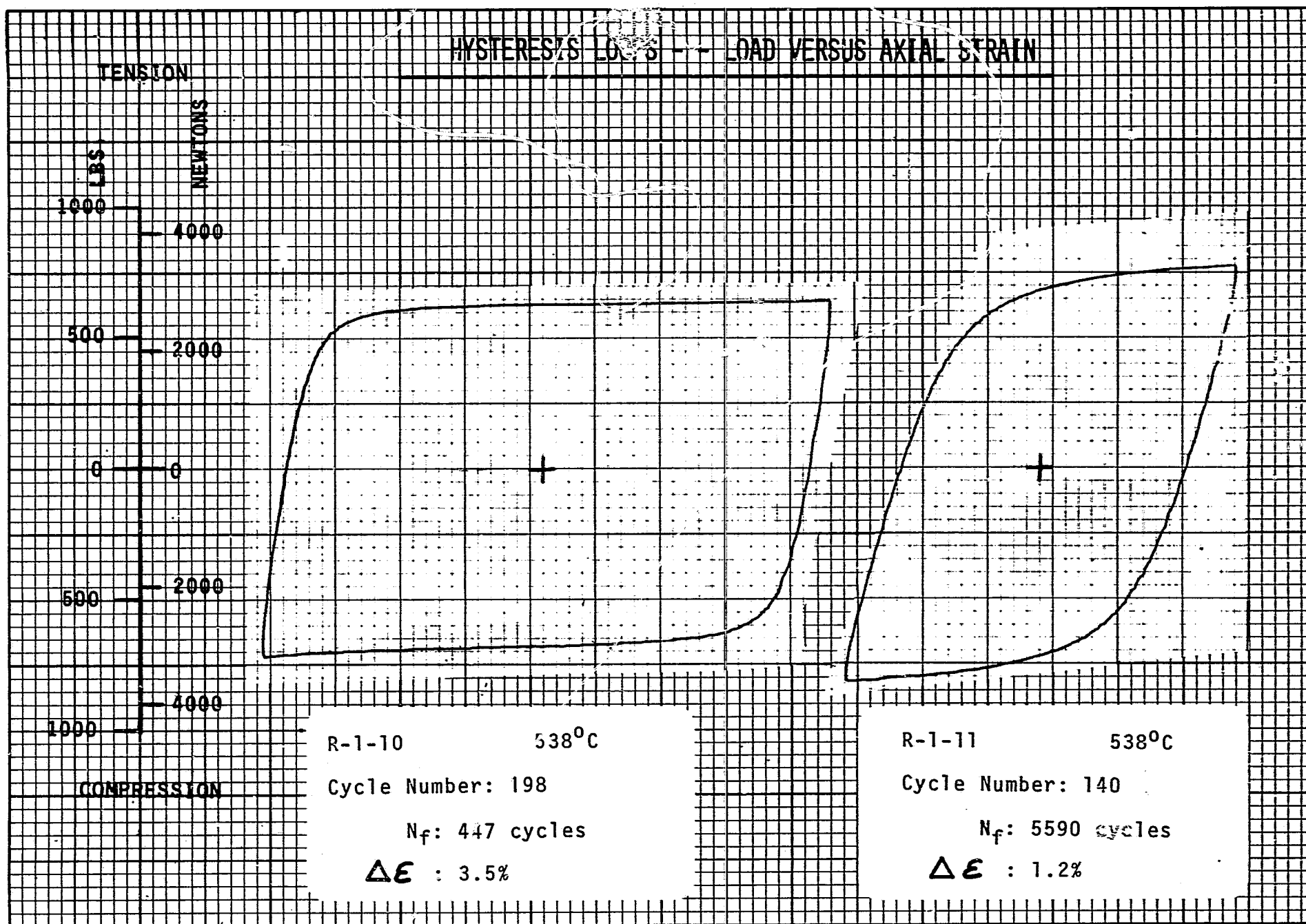
R-1-9

538°C

Cycle Number: 64

N_f : 562 cycles

ΔE : 3.5%



HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

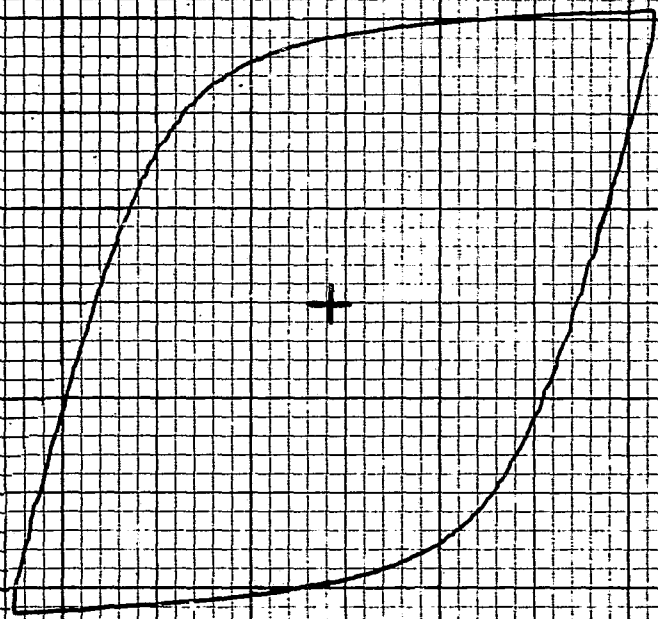
TENSION

1000
500
0
500
1000

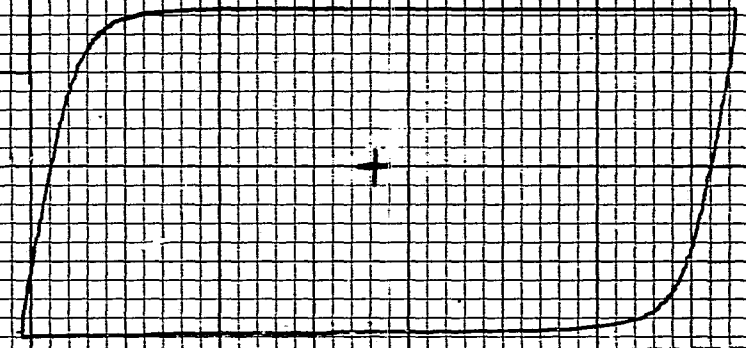
LB
NEWTONS

4000
2000
0
2000
4000

COMPRESSION



R-1-12 538°C
 Cycle Number: 218
 N_f: 3660 cycles
 Δε : 1.35%



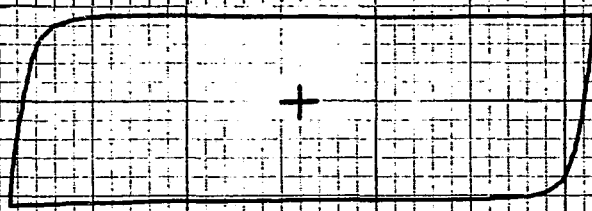
R-2-2 538°C
 Cycle Number: 760
 N_f: 1615 cycles
 Δε : 3.0%

Figure 196

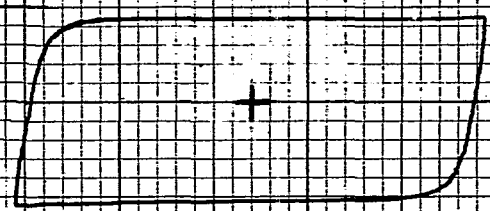
HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION
 2000
 1000
 0
 1000
 2000
 COMPRESSION

NEWTONS



R-2-4 538°C
 Cycle Number: 156
 N_f : 366 cycles
 $\Delta \epsilon$: 5.0%

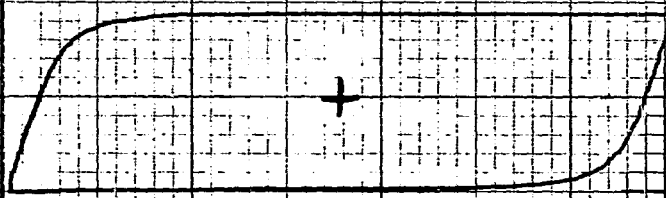


R-2-5 538°C
 Cycle Number: 183
 N_f : 552 cycles
 $\Delta \epsilon$: 4.0%

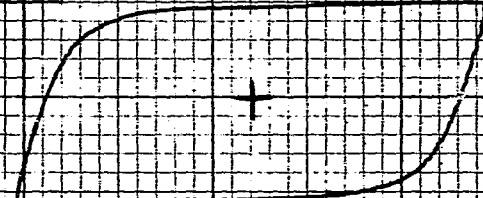
HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

LBS TENSION
2000
1000
0
1000
2000
COMPRESSION

NEWTONS
8000
4000
0
4000
8000



R-2-10 538°C
Cycle Number: 321
N_f: 1055 cycles
 ΔE : 2.8%



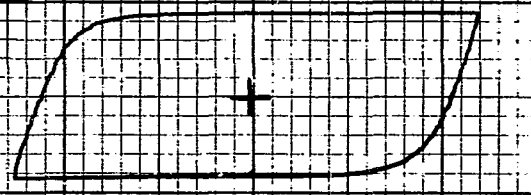
R-2-11 538°C
Cycle Number: 222
N_f: 1239 cycles
 ΔE : 2.0%

Figure 198

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

LIBS. TENSION
2000
1000
0
1000
2000
COMPRESSION

NEWTONS
8000
4000
0
4000
8000



R-2-6 538^oC
Cycle Number: 720
N_f: 2051 cycles
 $\Delta\epsilon$: 2.0%

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

LBS

NEWTONS

1000

500

0

500

1000

4000

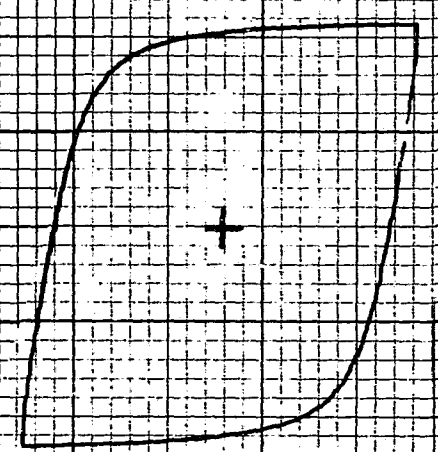
2000

0

2000

4000

COMPRESSION

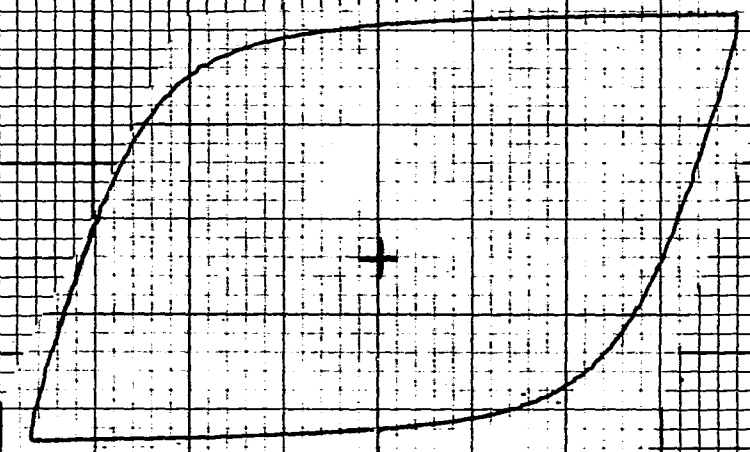


R-2-12 538°C

Cycle Number: 292

N_f : 1770 cycles

$\Delta\epsilon$: 1.7%



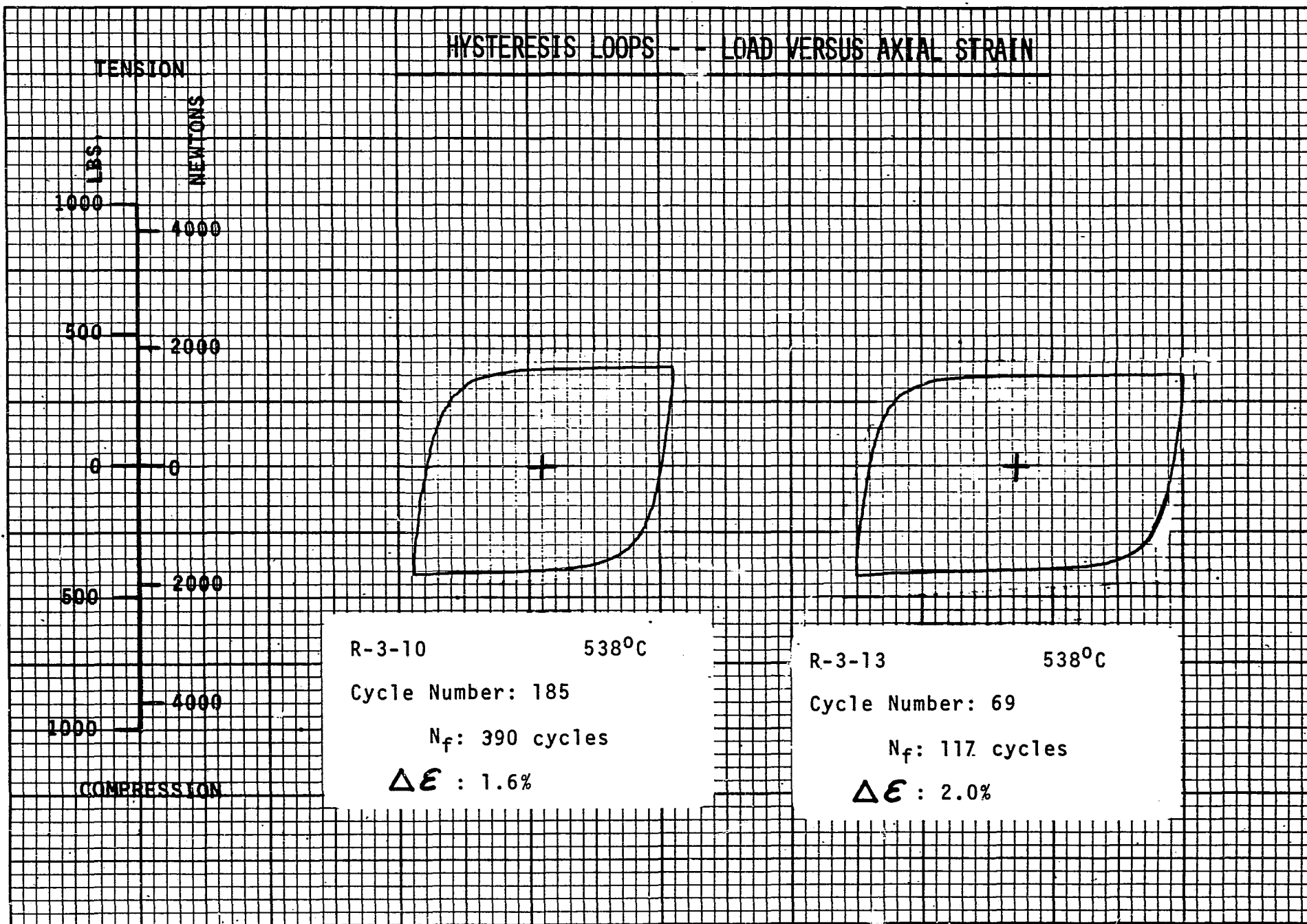
R-2-13 538°C

Cycle Number: 508

N_f : 2453 cycles

$\Delta\epsilon$: 1.5%

Figure 200



HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION

LBS

NEWTONS

1000

500

0

500

1000

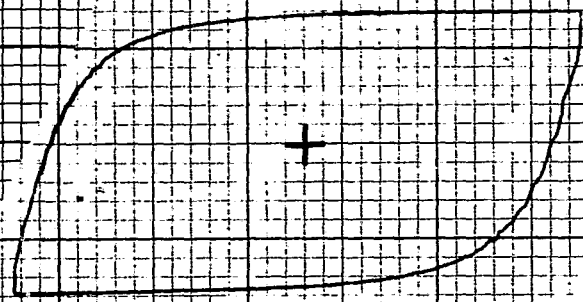
4000

2000

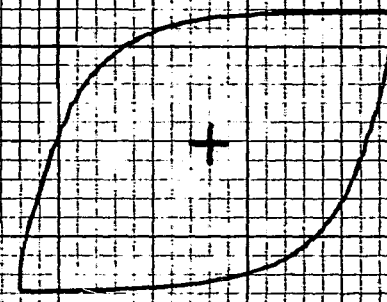
2000

4000

COMPRESSION

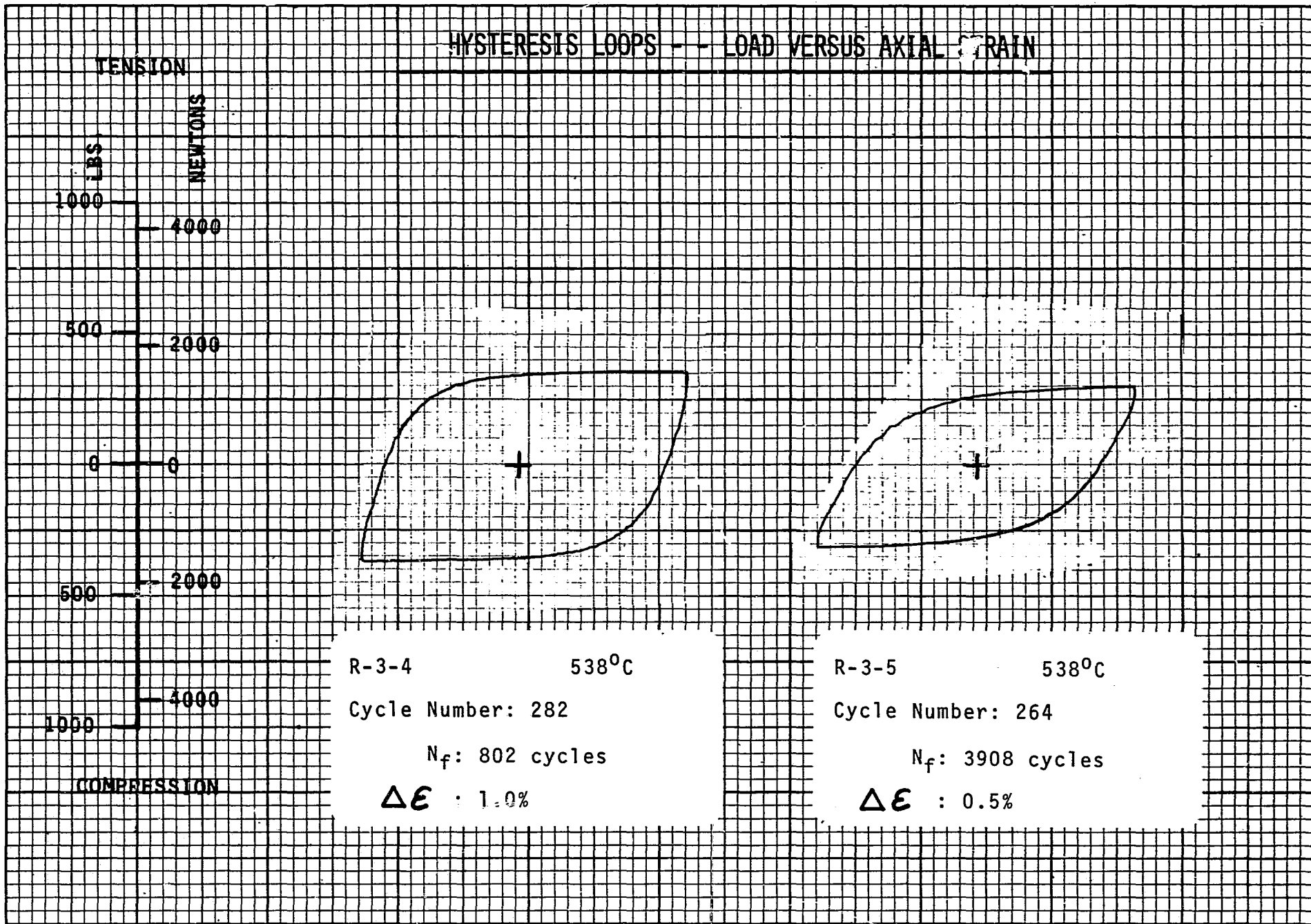


R-3-1 538°C
Cycle Number: 245
N_f: 462 cycles
 $\Delta\epsilon$: 1.2%



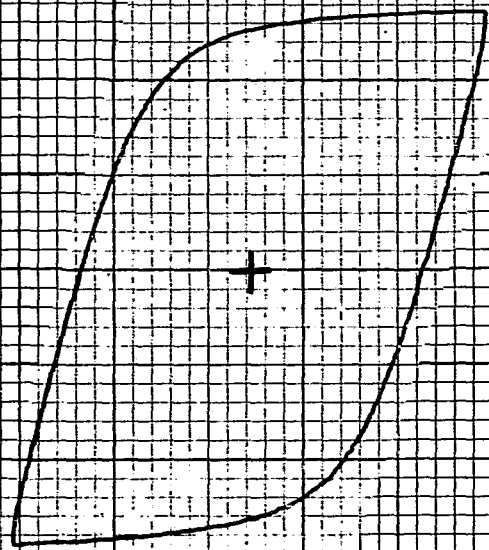
R-3-2 538°C
Cycle Number: 638
N_f: 1179 cycles
 $\Delta\epsilon$: 0.8%

Figure 202

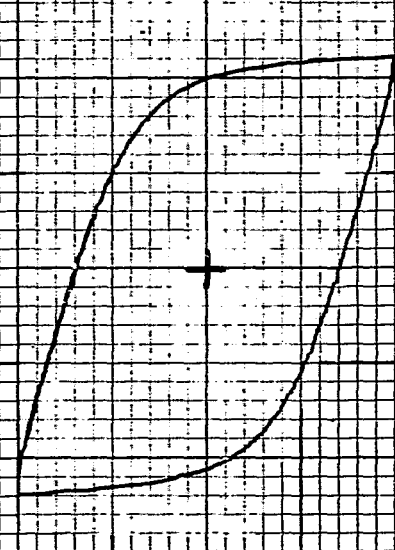


HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION
 LBS. NEWTONS
 2000 8000
 1000 4000
 0 0
 1000 4000
 2000 8000
 COMPRESSION



R-4-1 538°C
 Cycle Number: 32
 N_f : 147 cycles
 $\Delta \epsilon$: 2.0%



R-4-2 538°C
 Cycle Number: 169
 N_f : 354 cycles
 $\Delta \epsilon$: 1.6%

Figure 204

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

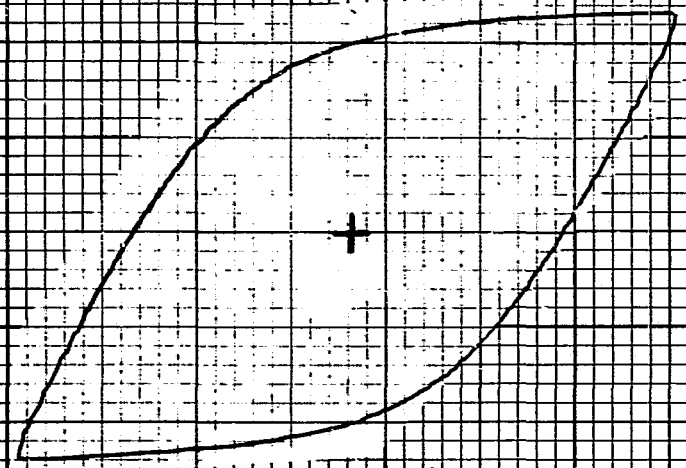
2000
1000
0
1000
2000

TENSION

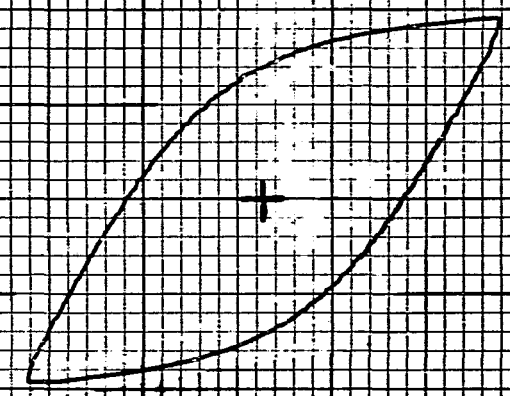
NEWTONS

8000
4000
0
4000
8000

COMPRESSION



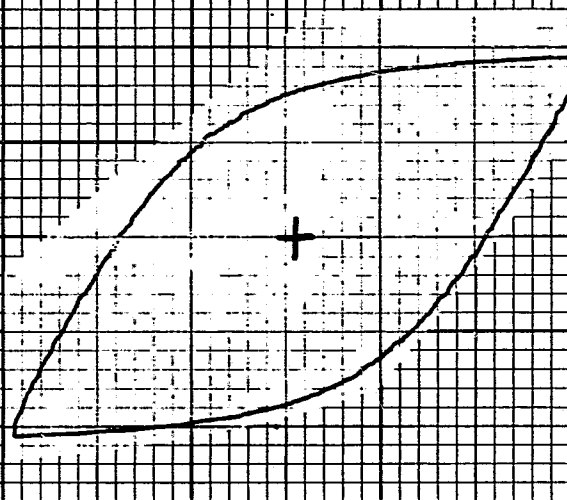
R-4-3 538°C
Cycle Number: 227
N_f: 605 cycles
 $\Delta\epsilon$: 1.4%



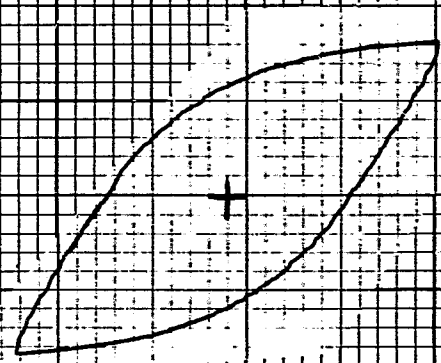
R-4-4 538°C
Cycle Number: 720
N_f: 1823 cycles
 $\Delta\epsilon$: 1.0%

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION
 LBS. 2000
 1000
 0
 1000
 2000
 NEWTONS 8000
 4000
 0
 4000
 8000
 COMPRESSION

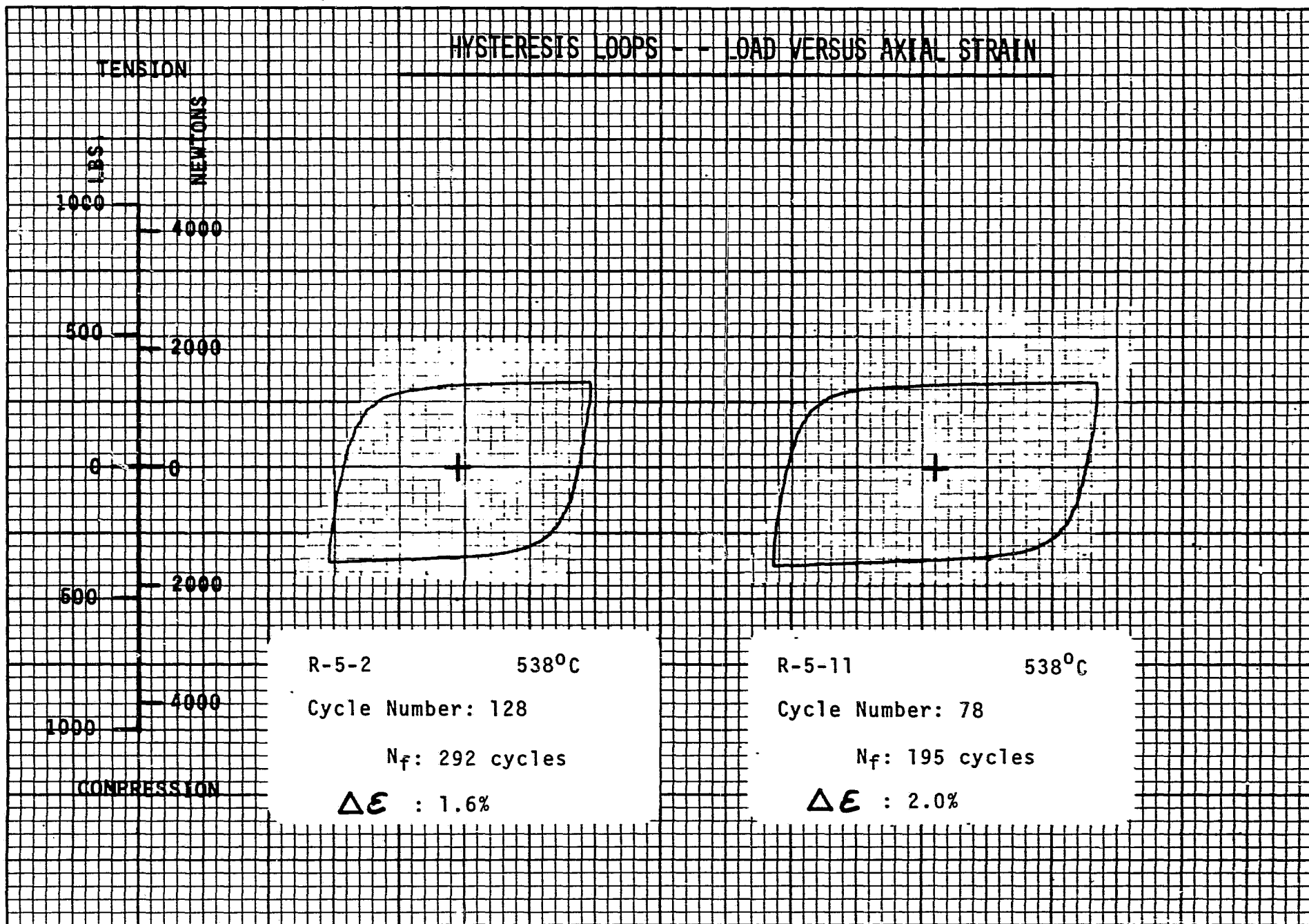


R-4-5 538°C
 Cycle Number: 545
 N_f : 1102 cycles
 $\Delta \epsilon$: 1.2%



R-4-8 538°C
 Cycle Number: 1748
 N_f : 3648 cycles
 $\Delta \epsilon$: 0.9%

Figure 206



HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

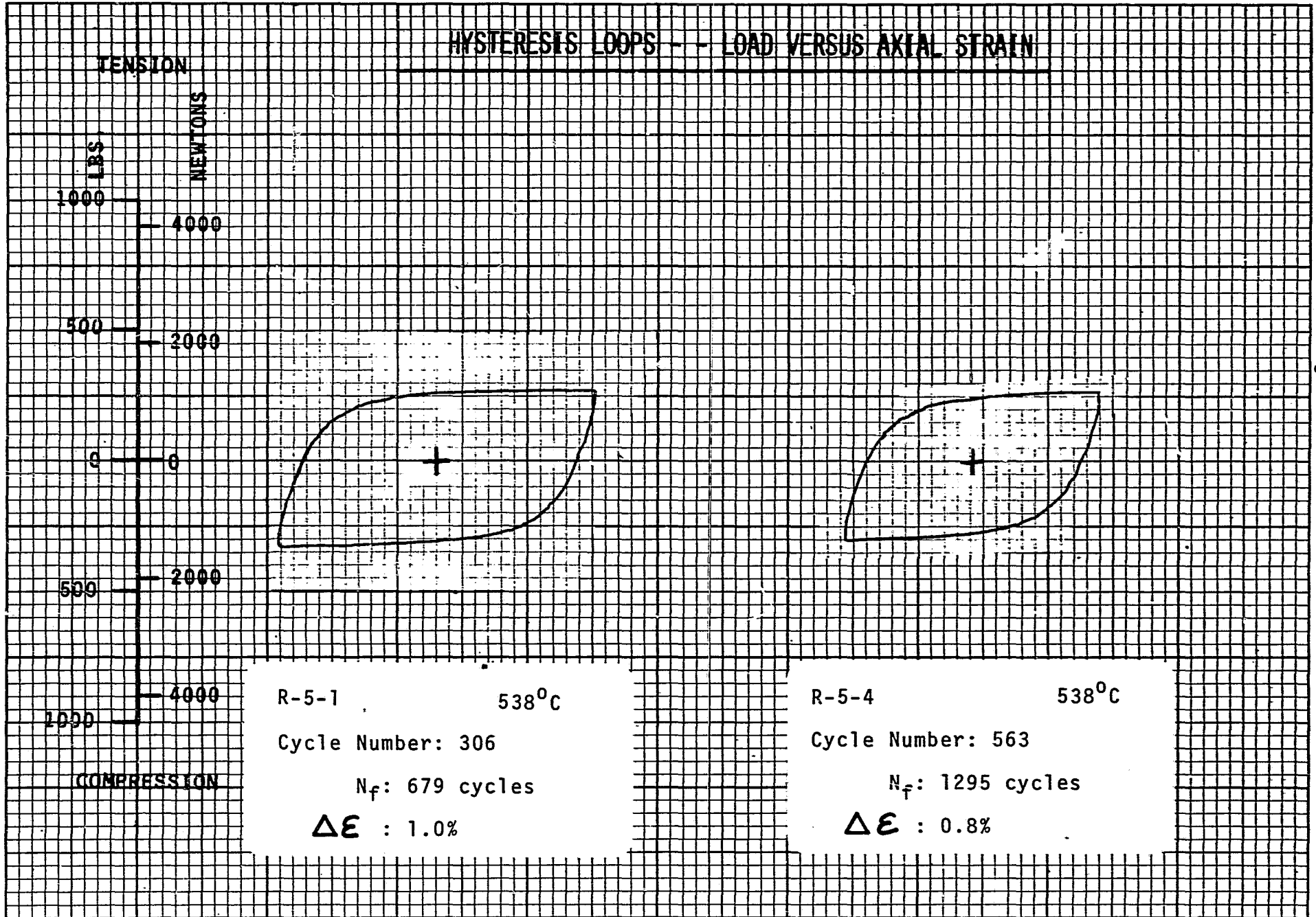
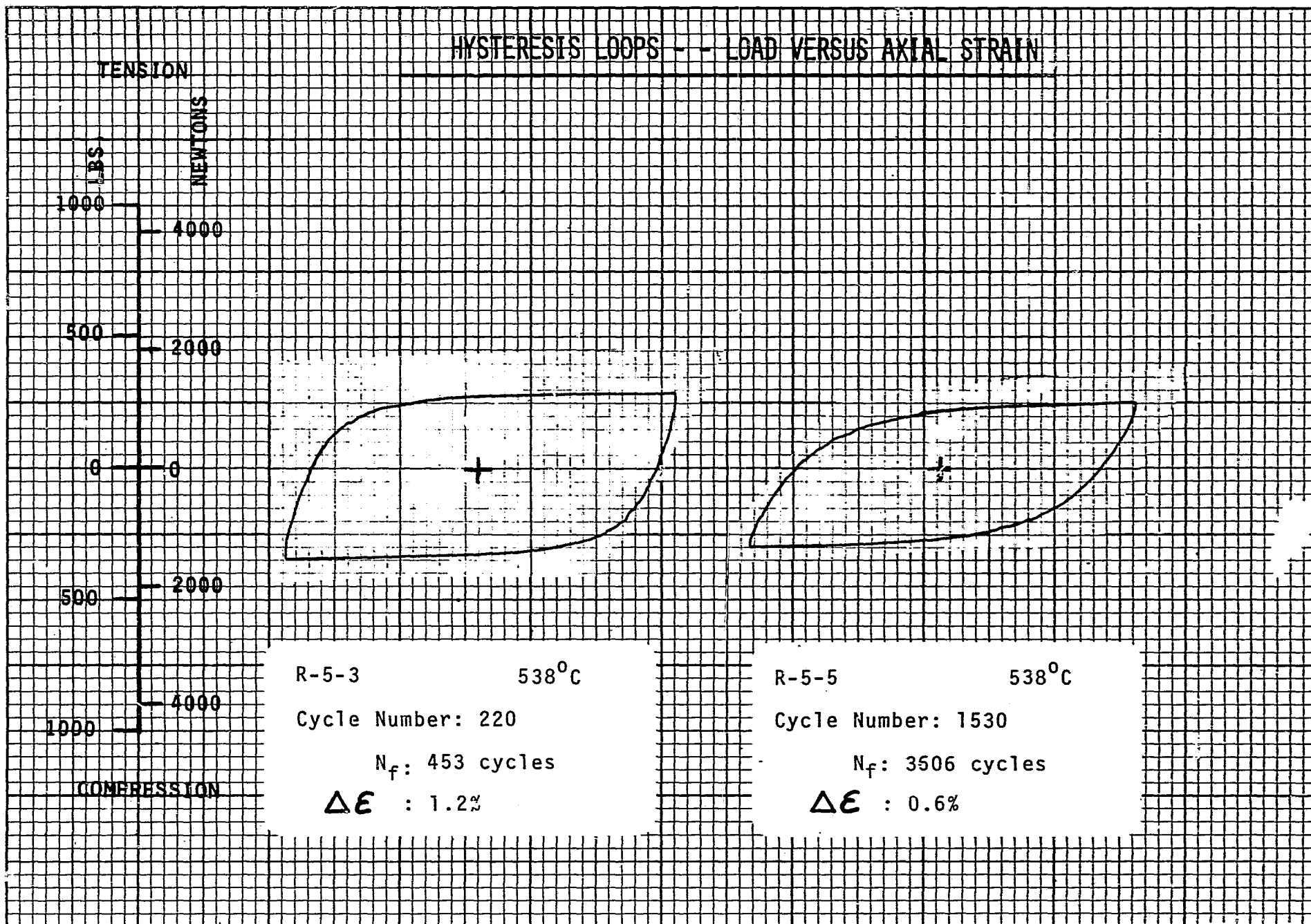


Figure 208



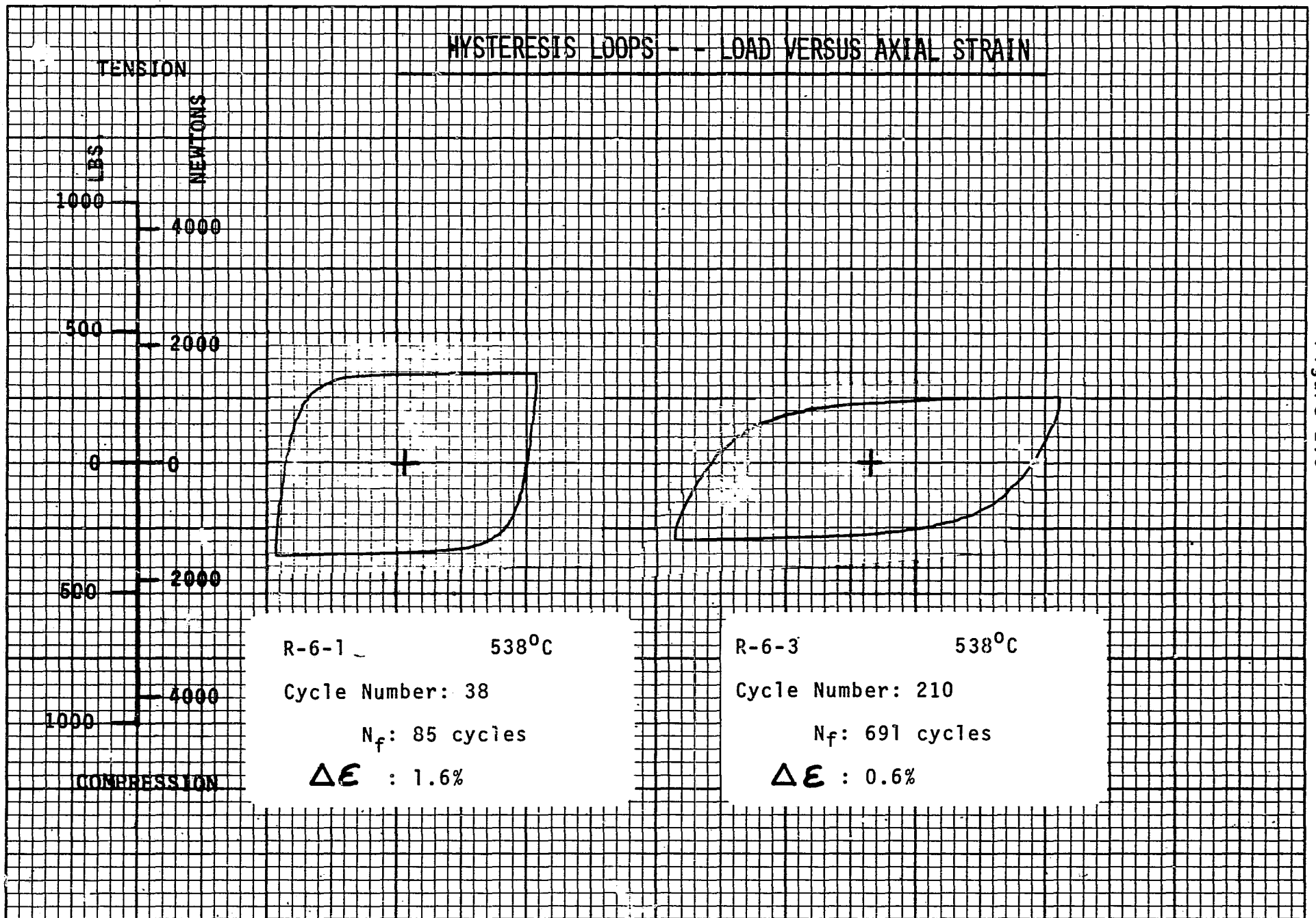
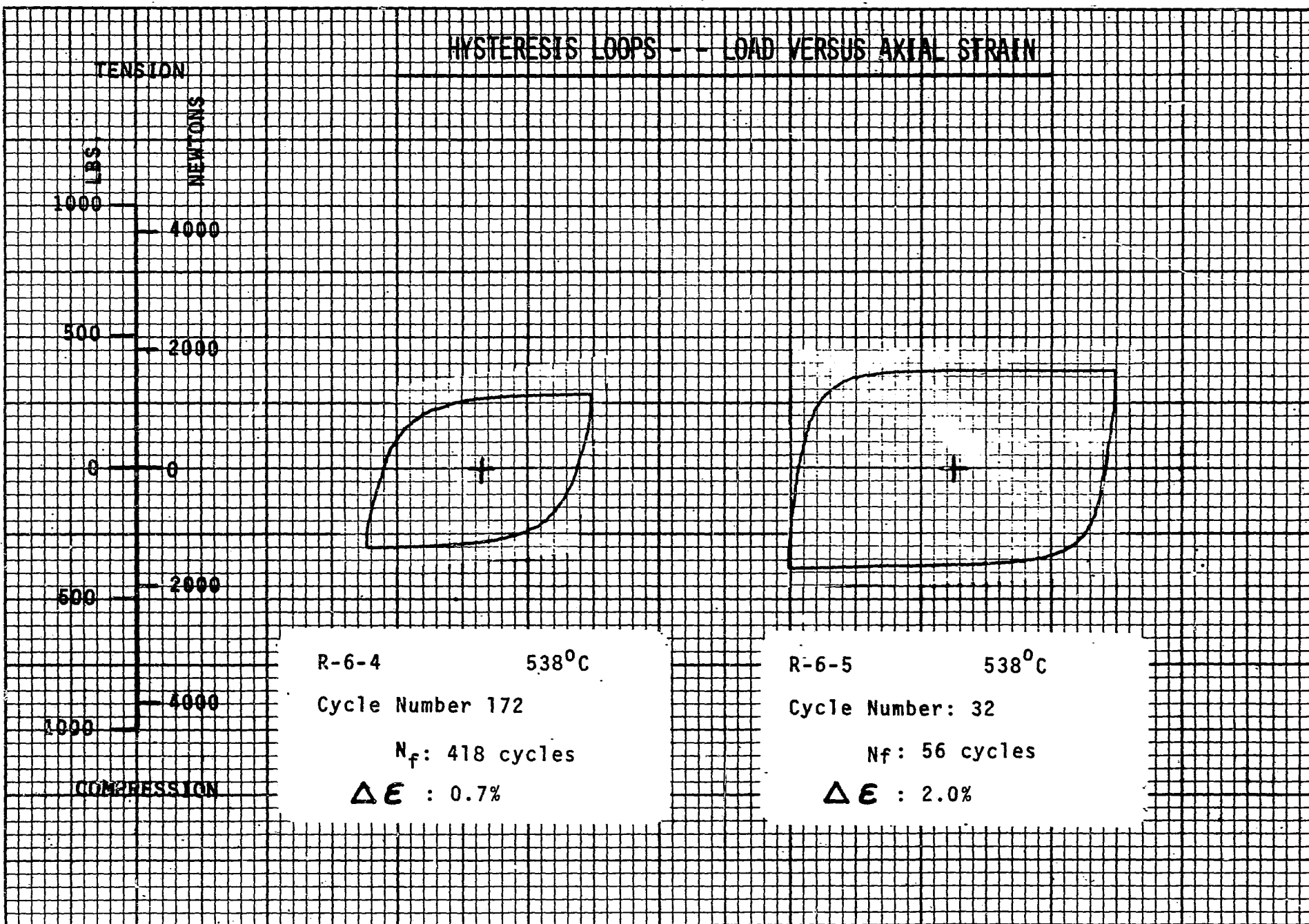


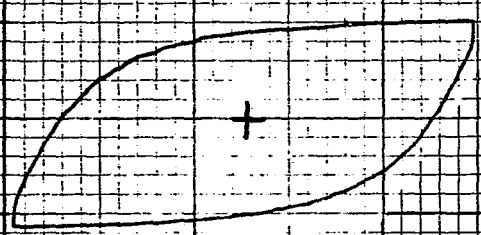
Figure 210

Figure 211

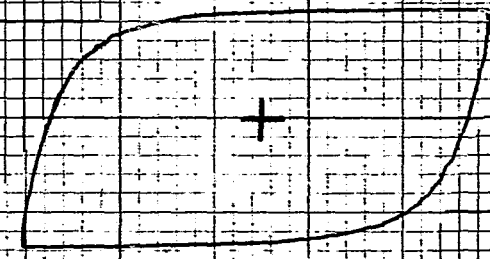


HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION
LBS
NEWTONS
1000
500
0
500
1000
COMPRESSION

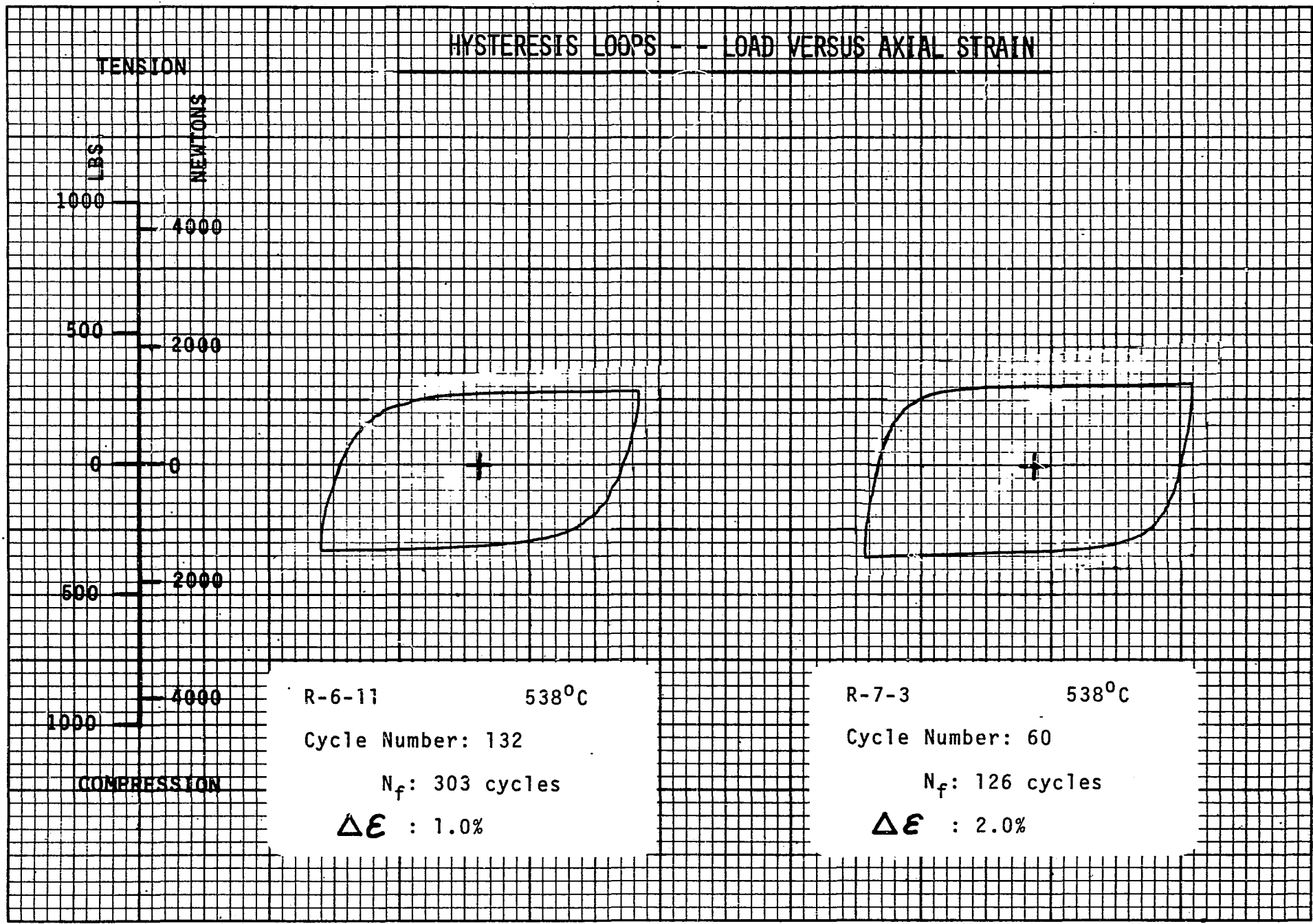


R-6-8 538^oC
Cycle Number: 600
N_f: 1358 cycles
 $\Delta \epsilon$: 0.5%



R-6-9 538^oC
Cycle Number: 60
N_f: 200 cycles
 $\Delta \epsilon$: 1.0%

Figure 212



TENSION

LBS

NEWTONS

1000

4000

500

2000

0

0

500

2000

1000

4000

COMPRESSION

R-6-11 538°C

Cycle Number: 132

N_f : 303 cycles

$\Delta \epsilon$: 1.0%

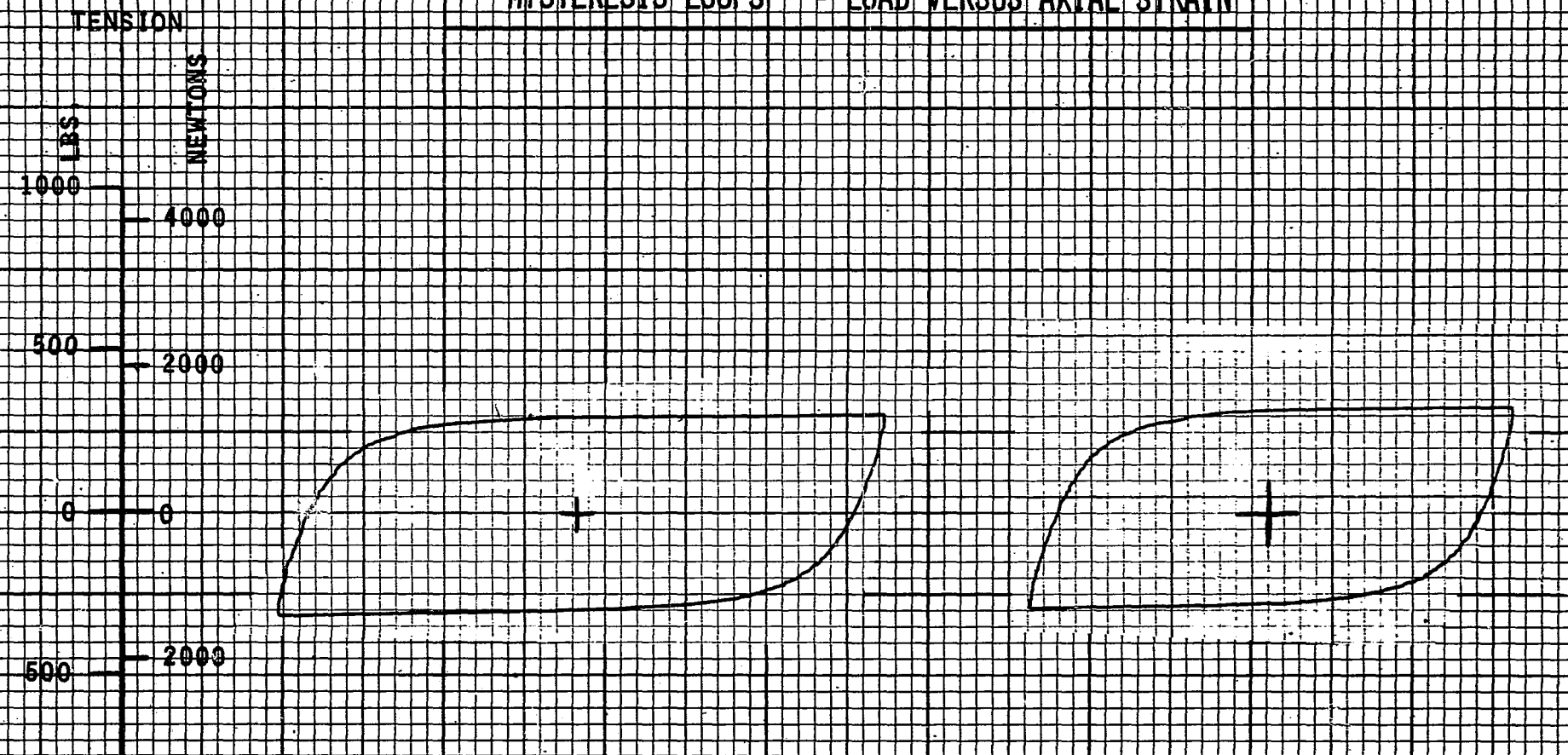
R-7-3 538°C

Cycle Number: 60

N_f : 126 cycles

$\Delta \epsilon$: 2.0%

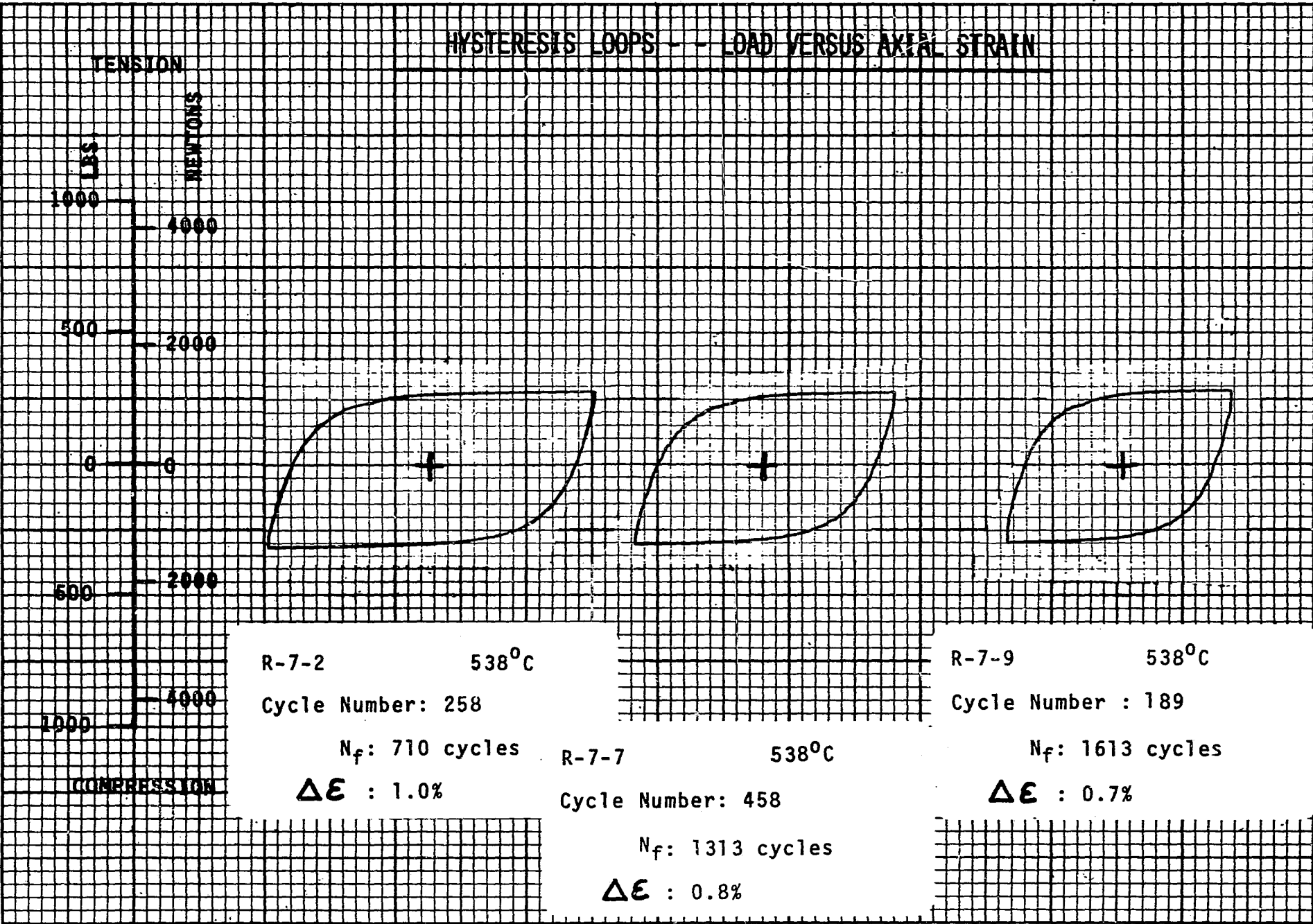
HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN



R-7-4 538°C
 Cycle Number: 120
 N_f : 269 cycles
 $\Delta \epsilon$: 1.5%

R-7-1 538°C
 Cycle Number : 240
 N_f : 437 cycles
 $\Delta \epsilon$: 1.2%

Figure 214



HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION

LBS

NEWTONS

1000

500

0

500

1000

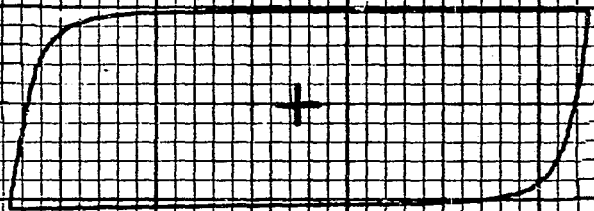
4000

2000

2000

4000

COMPRESSION



R-8-7 538°C

Cycle Number: 26

N_f : 344 cycles

ΔE : 3.0%

Figure 216

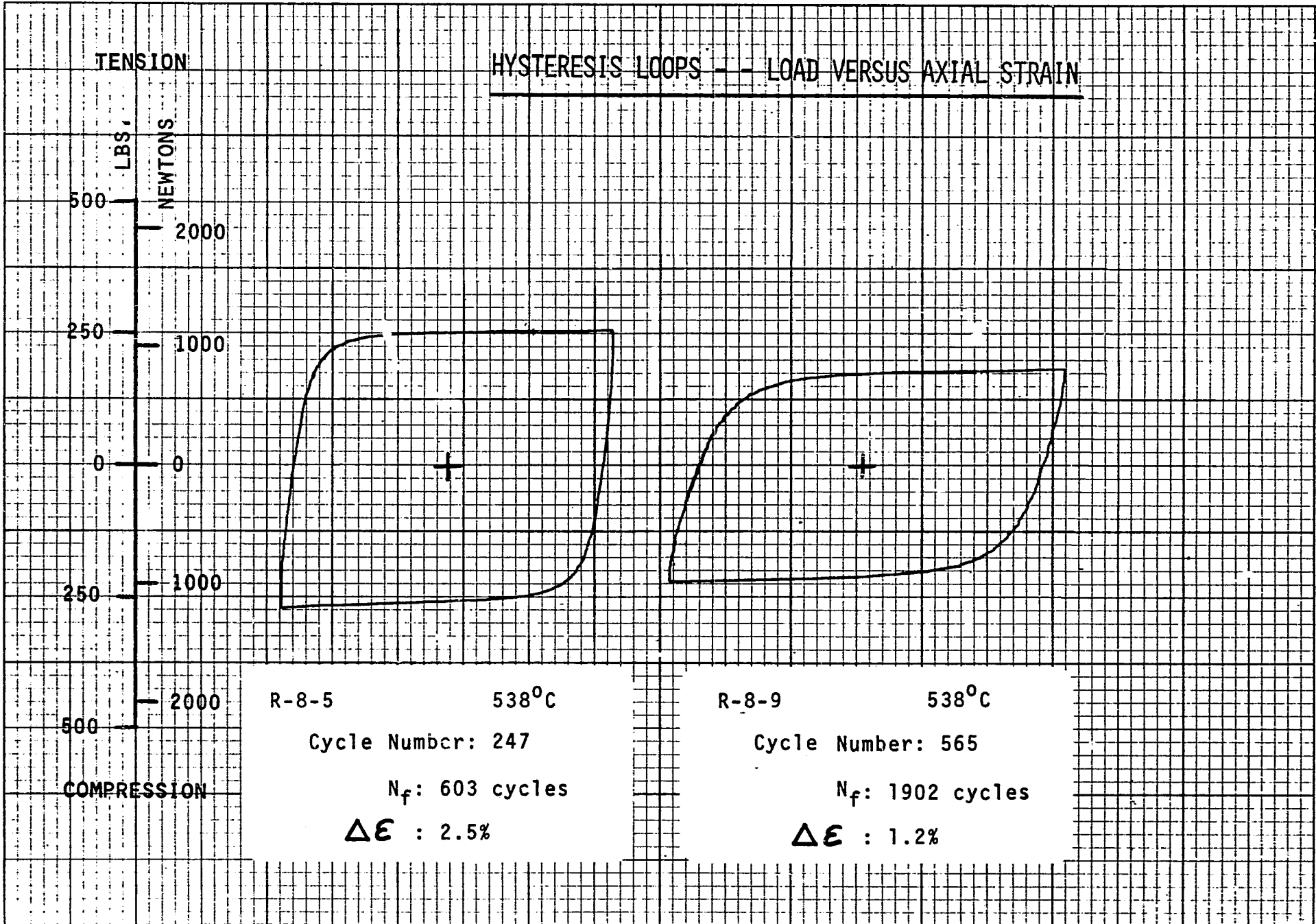


Figure 217

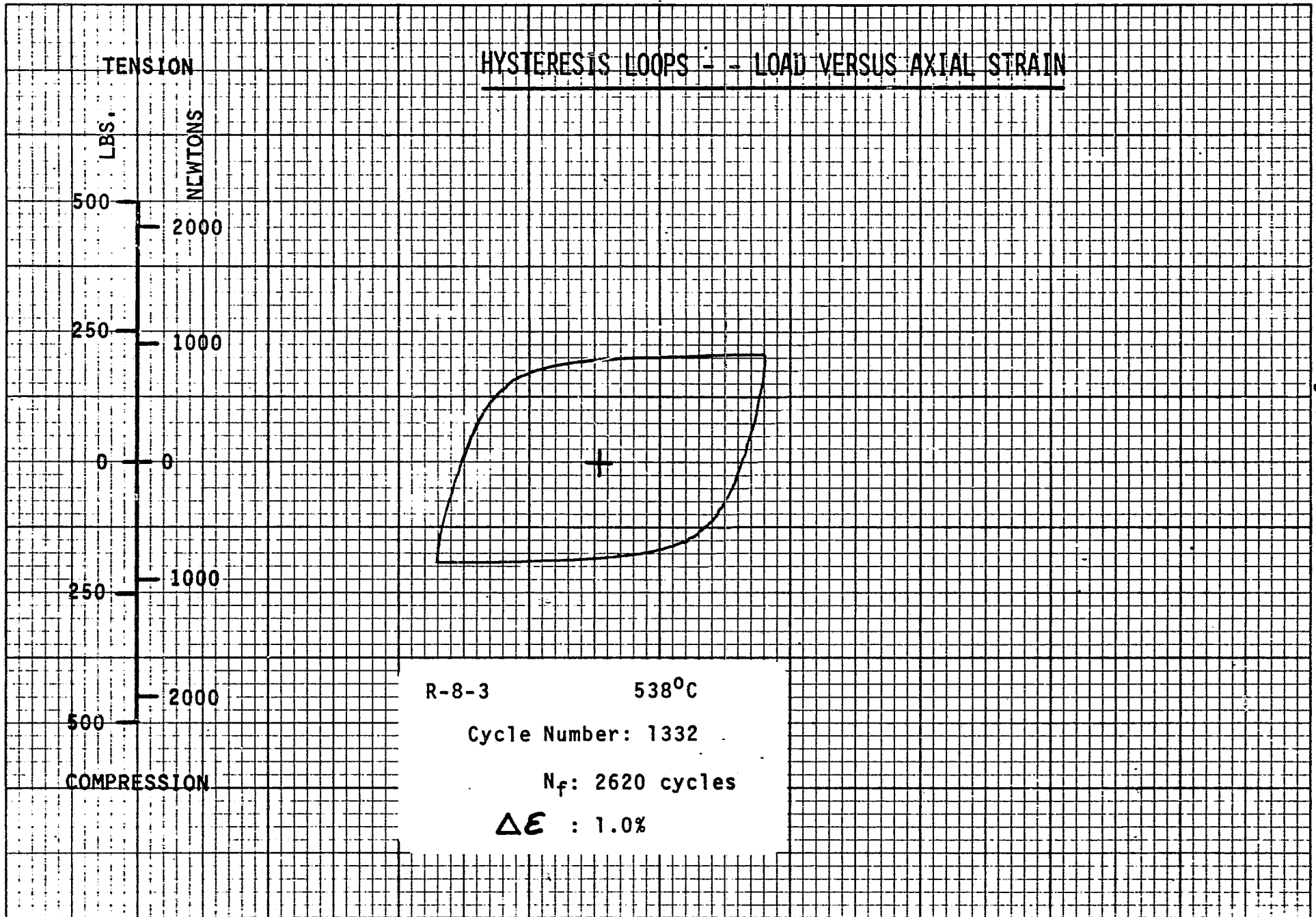
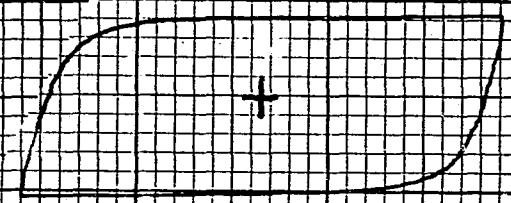


Figure 218

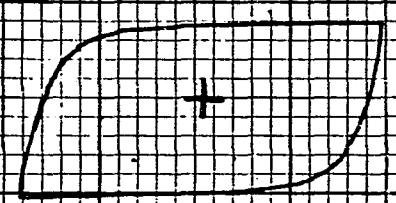
HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION
1000 LBS
500
0
500
1000
COMPRESSION

NEWTONS
4000
2000
0
2000
4000



R-8-2 538°C
Cycle Number: 369
N_f: 928 cycles
 $\Delta \epsilon$: 2.0%



R-8-10 -538°C
Cycle Number: 680
N_f: 1381 cycles
 $\Delta \epsilon$: 1.5%

Figure 219

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION
 LBS
 NEWTONS

2000
 1000
 0
 1000
 2000

R-9-11 538^oC

Cycle Number: 362

N_f : 843 cycles

$\Delta \epsilon$: 2.0%

R-9-4 538^oC

Cycle Number: 178

N_f : 357 cycles

$\Delta \epsilon$: 3.0%

COMPRESSION

Figure 220

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

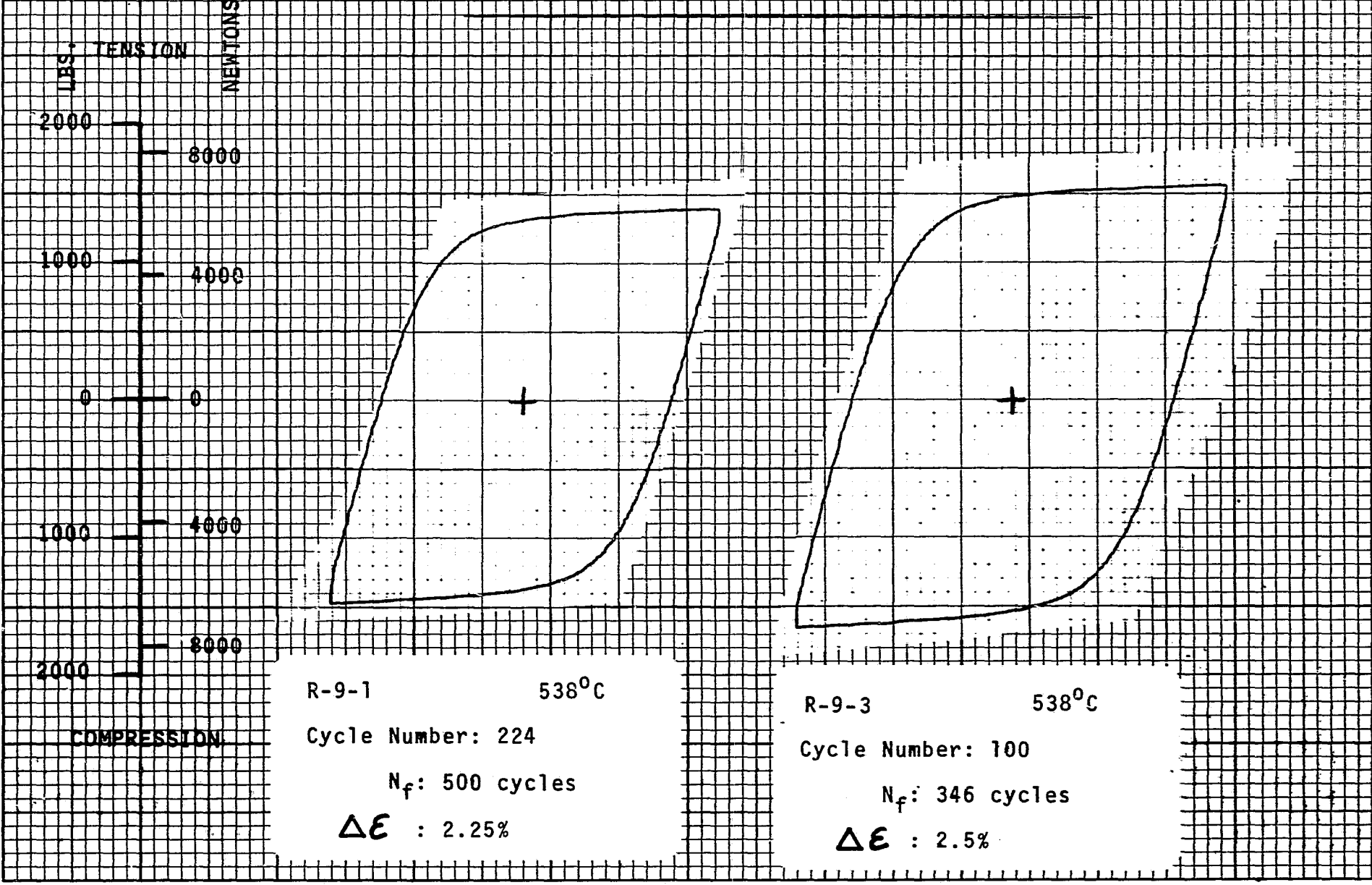


Figure 221

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

LBS TENSION
NEWTONS

2000

8000

1000

4000

0

0

+

1000

4000

2000

8000

COMPRESSION

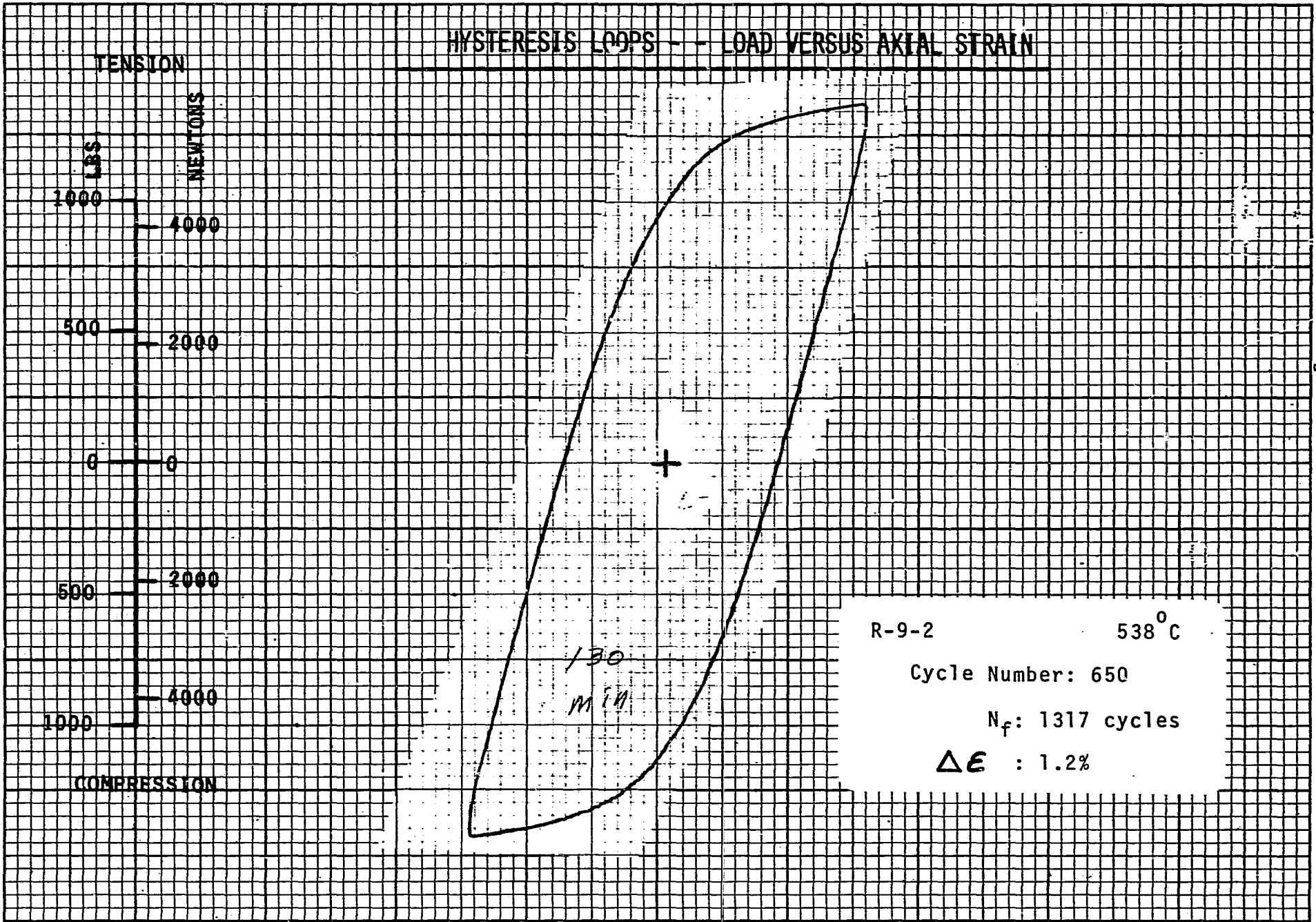
R-9-8

538°C

Cycle Number: 313

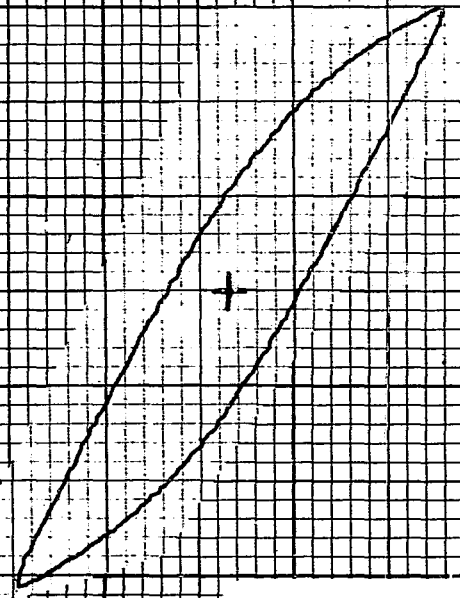
N_f : 2000 cycles

$\Delta \epsilon$: 1.4%



HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

LIBS TENSION
2000
1000
0
1000
2000
NEWTONS
8000
4000
0
4000
8000
COMPRESSION



R-9-9 538°C
Cycle Number: 227
N_f: 6670 cycles
 $\Delta\epsilon$: 0.9%

Figure 224

Figure 225

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION

1000
500
0
500
1000

NEWTONS

4000
2000
0
2000
4000

1000

COMPRESSION

R-10-3 538°C

Cycle Number : 56

N_f : 148 cycles

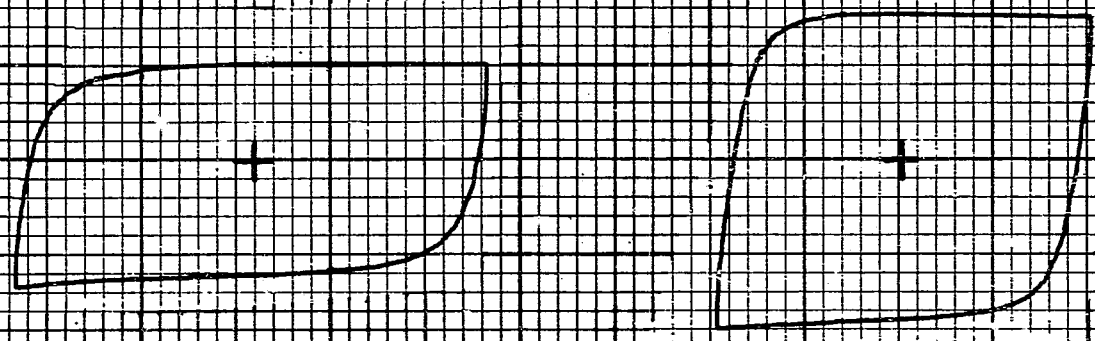
ΔE : 2.0%

R-10-4 538°C

Cycle Number : 7

N_f : 38 cycles

ΔE : 1.6%



HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

NEWTONS

1000

500

0

500

1000

4000

2000

2000

4000

1000

COMPRESSION

R-10-1

538^oC

Cycle Number: 678

N_f : 1542 cycles

$\Delta \epsilon$: 0.8%

R-10-8

538^oC

Cycle Number: 9

N_f : 72 cycles

$\Delta \epsilon$: 1.2%

Figure 226

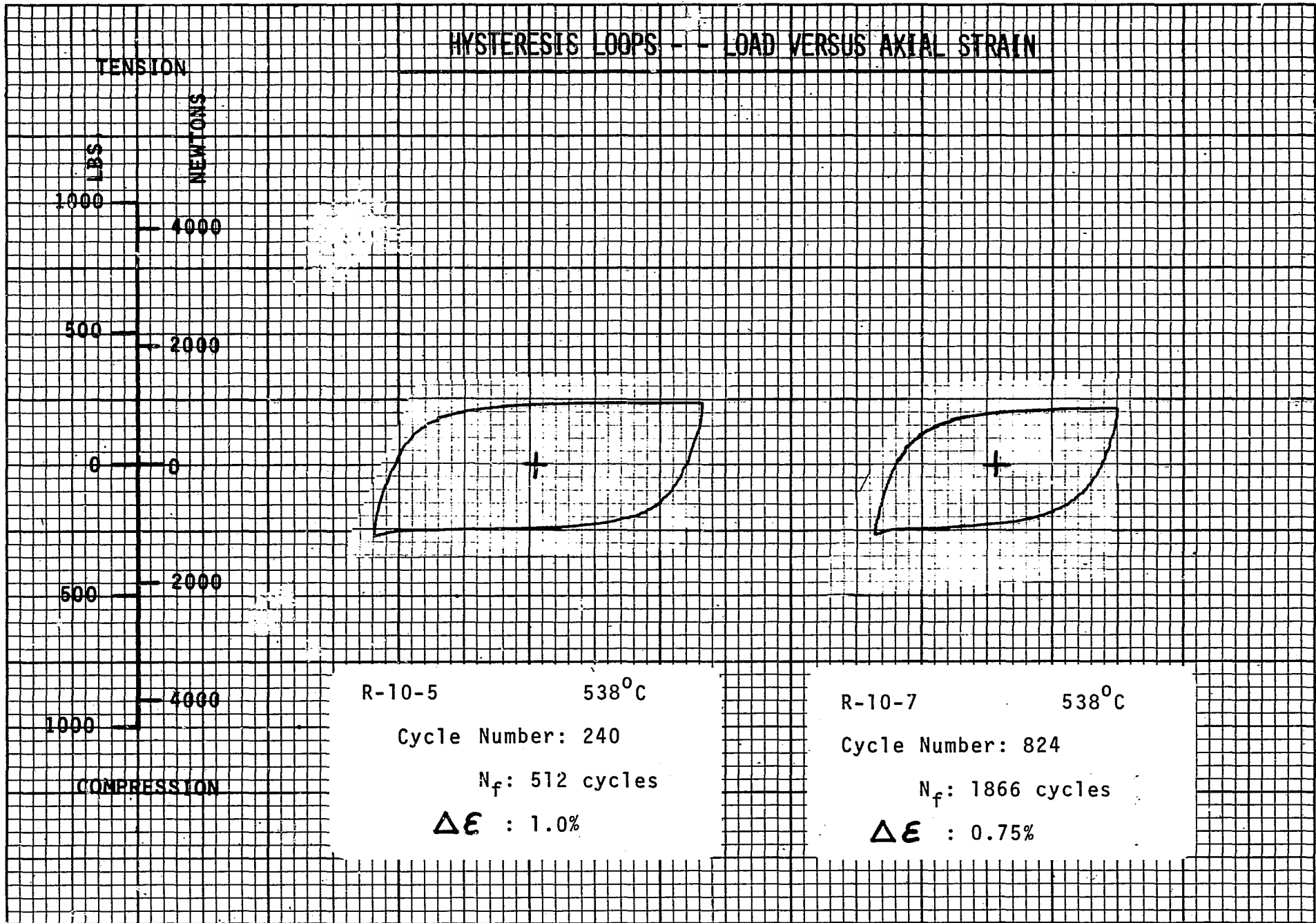


Figure 227

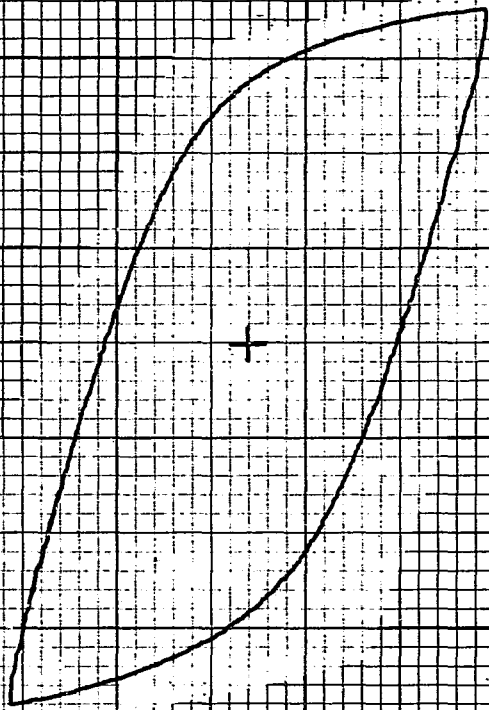
HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

LBS. TENSION

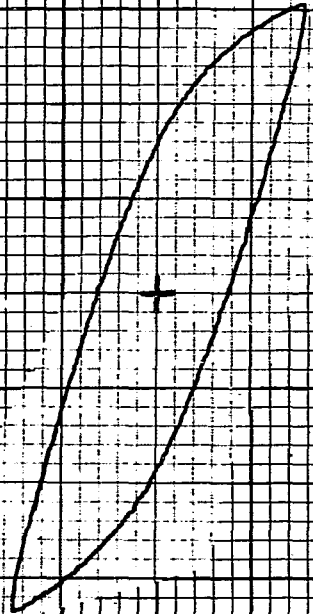
NEWTONS

2000
1000
0
1000
2000

COMPRESSION

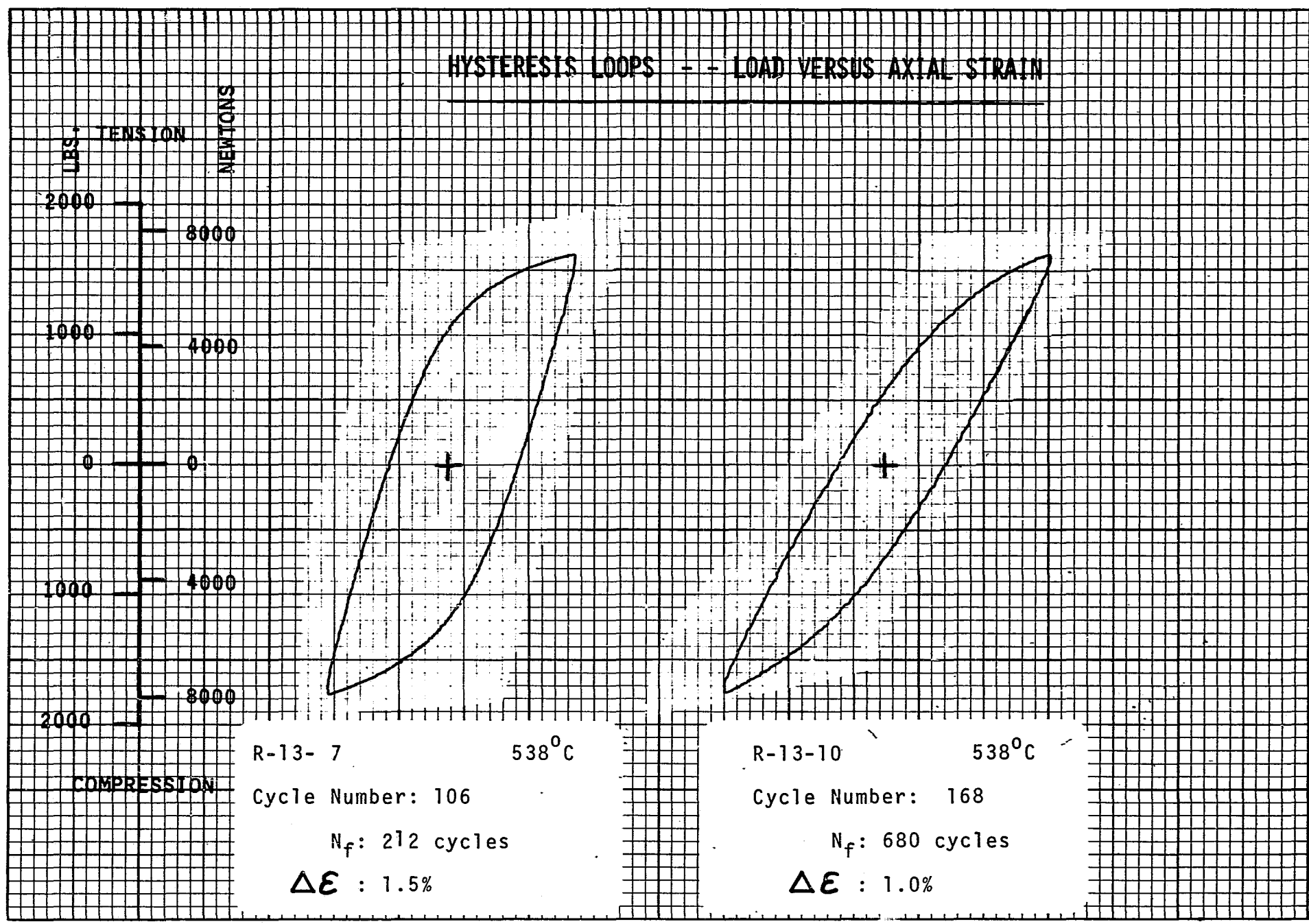


R-13-5 538⁰C
 Cycle Number: 45
 N_f: 90 cycles
 Δε : 2.0%



R-13-13 538⁰C
 Cycle Number: 325
 N_f: 644 cycles
 Δε : 1.2%

Figure 228



HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

LIB. TENSION
NEWTONS

2000

8000

1000

4000

0

0

1000

4000

2000

8000

COMPRESSION

R-13-6

538°C

Cycle Number: 480

N_f : 1615 cycles

ΔE : 0.8%

R-13-8

538°C

Cycle Number: 515

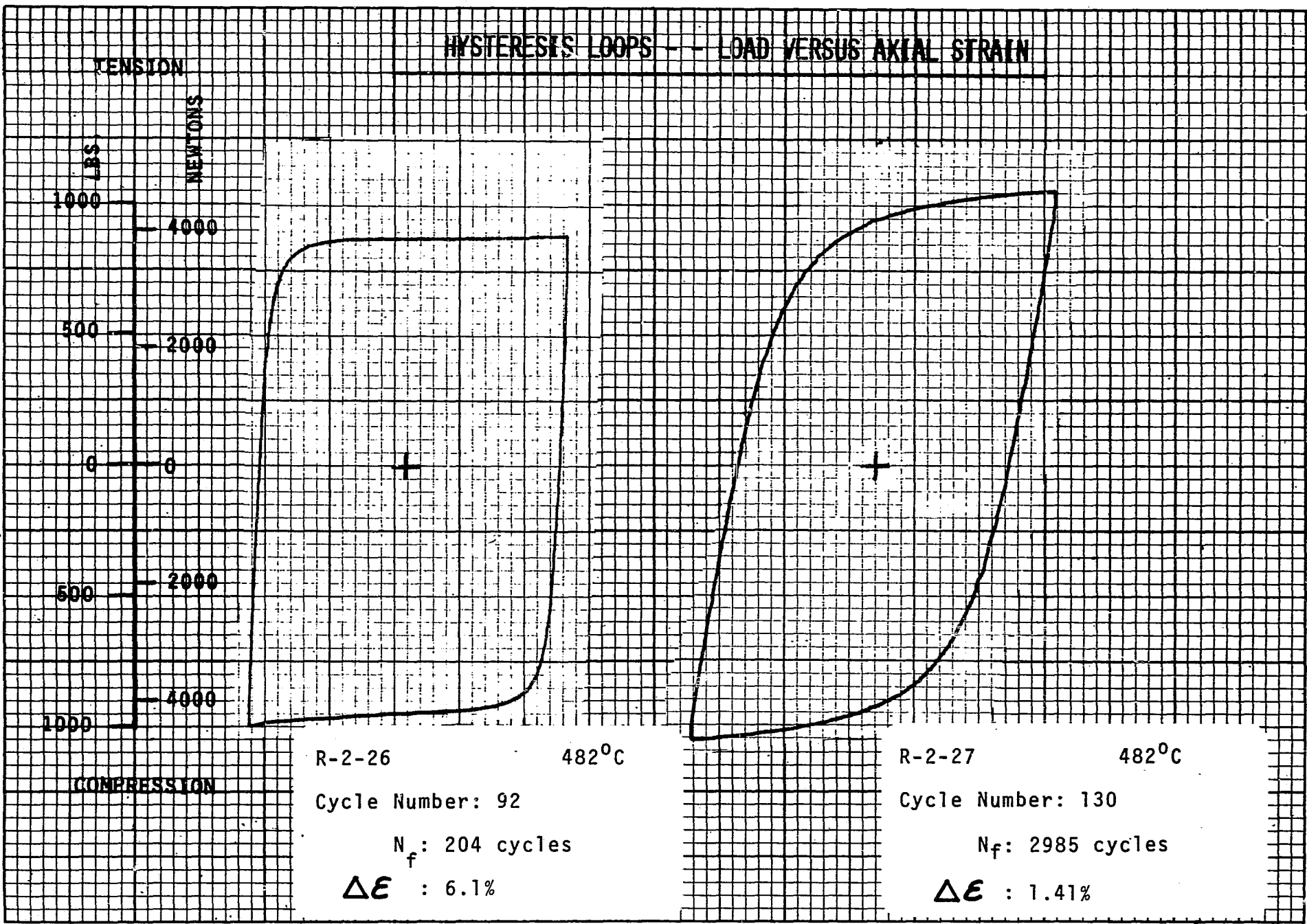
N_f : 3623 cycles

ΔE : 0.7%

Figure 230

Figure 231

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN



TENSION

NEWTONS

1000
500
0
500
1000

4000
2000
0
2000
4000

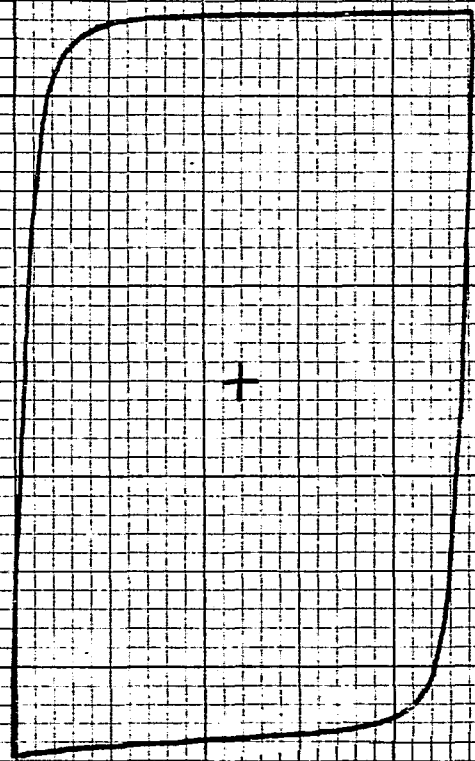
COMPRESSION

R-2-26 482°C
 Cycle Number: 92
 N_f : 204 cycles
 ΔE : 6.1%

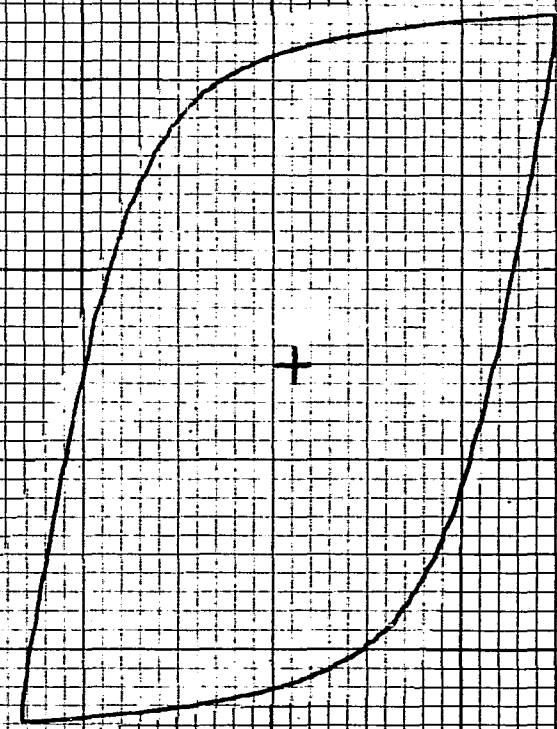
R-2-27 482°C
 Cycle Number: 130
 N_f : 2985 cycles
 ΔE : 1.41%

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION
LBS
NEWTONS
1000
500
0
500
1000
4000
2000
0
2000
4000



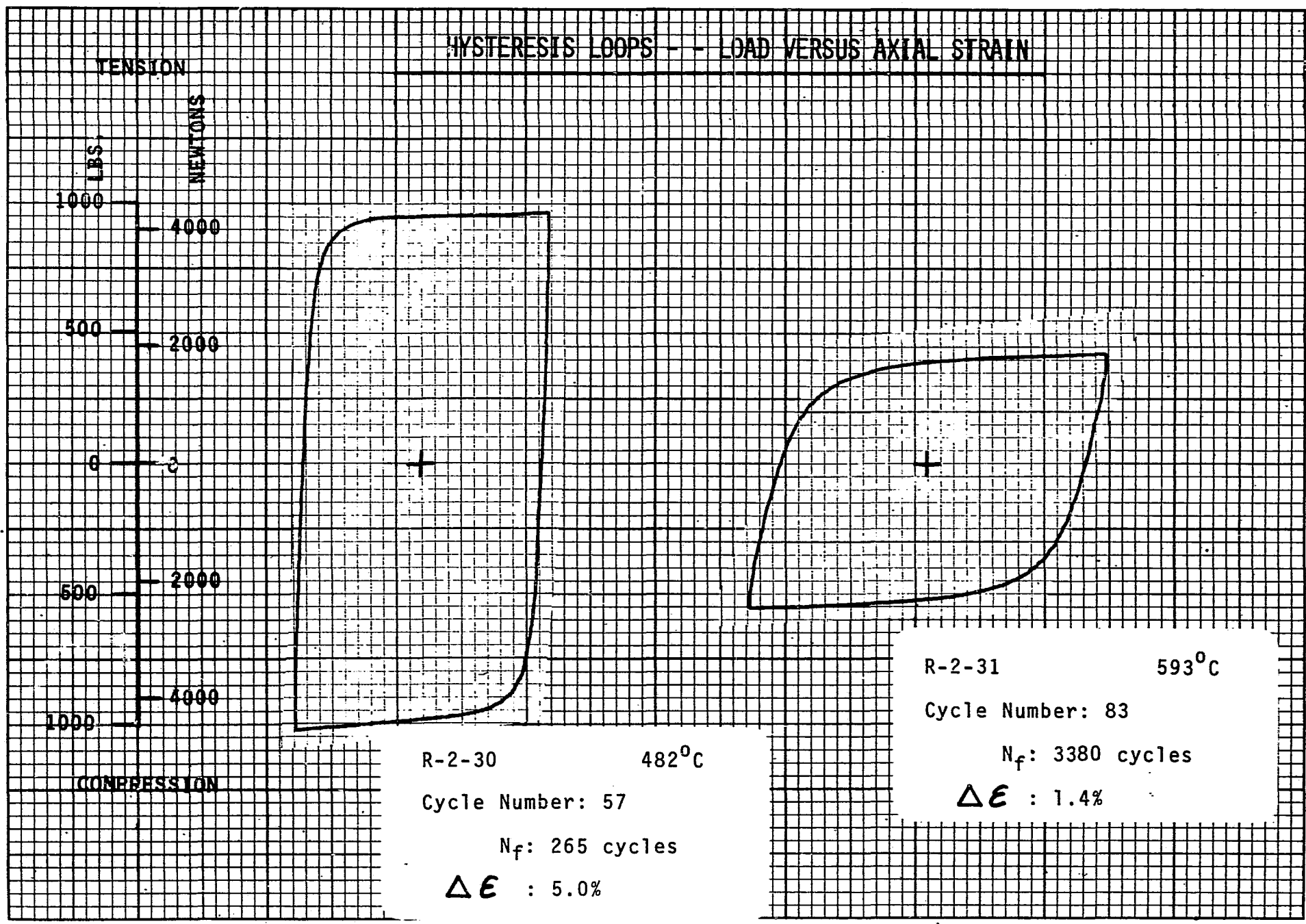
R-2-28 482°C
Cycle Number: 78
 N_f : 176 cycles
 ΔE : 6.1%



R-2-29 482°C
Cycle Number: 344
 N_f : 2135 cycles
 ΔE : 1.41%

Figure 232

Figure 233



R-2-31 593^oC
Cycle Number: 83
 N_f : 3380 cycles
 $\Delta \epsilon$: 1.4%

R-2-30 482^oC
Cycle Number: 57
 N_f : 265 cycles
 $\Delta \epsilon$: 5.0%

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

LBS

NEWTONS

1000

500

0

500

1000

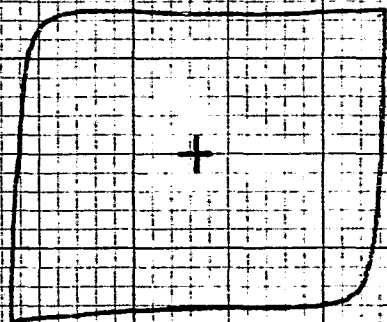
4000

2000

0

2000

4000

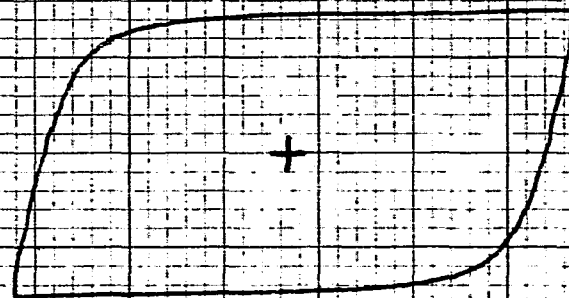


R-2-32 593°C

Cycle Number: 154

N_f : 346 cycles

$\Delta \epsilon$: 5.0%



R-2-33 593°C

Cycle Number: 490

N_f : 2008 cycles

$\Delta \epsilon$: 1.5%

COMPRESSION

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION

LBS

NEWTONS

1000

500

0

500

1000

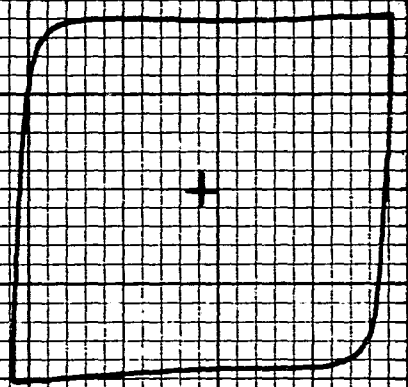
4000

2000

0

2000

4000

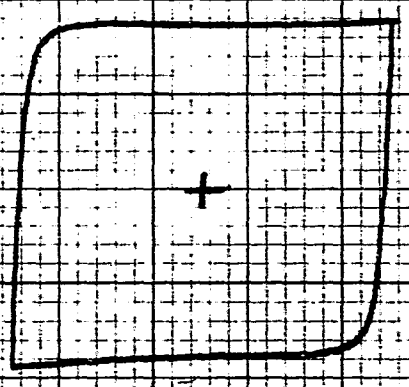


R-2-34 593°C

Cycle Number: 53

N_f : 234 cycles

$\Delta \epsilon$: 5.0%



R-2-36 538°C

Cycle Number: 47

N_f : 234 cycles

$\Delta \epsilon$: 5.0%

COMPRESSION

Figure 235

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

LBS NEWTONS

1000 4000

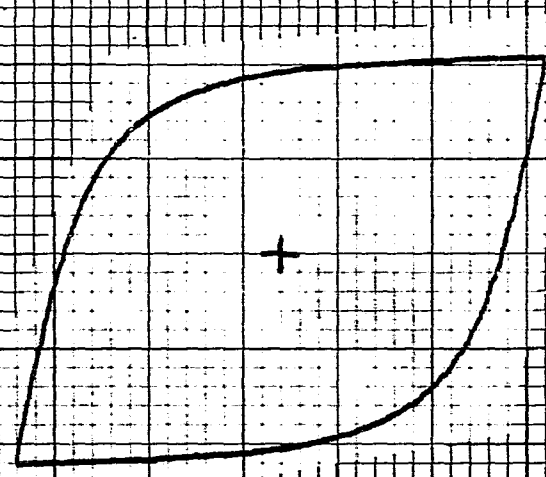
500 2000

0 0

500 2000

1000 4000

COMPRESSION

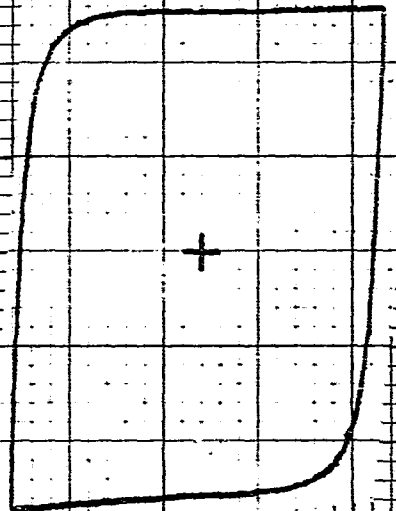


R-2-37 538^oC

Cycle Number: 60

N_f : 1613 cycles

$\Delta \epsilon$: 1.4%



R-2-39 538^oC

Cycle Number: 10

N_f : 238 cycles

$\Delta \epsilon$: 5.0%

Figure 236

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

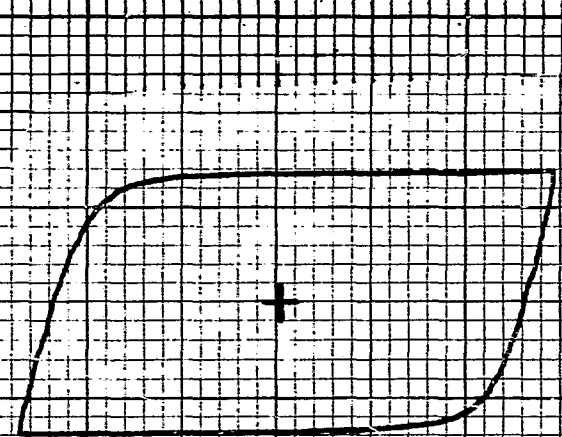
TENSION

LBS
NEWTONS

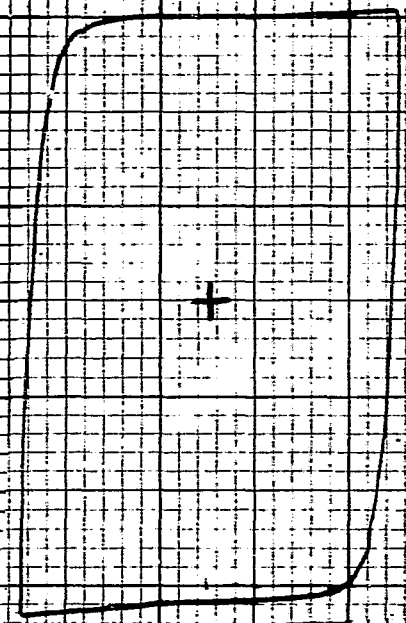
1000
500
0
500
1000

4000
2000
0
2000
4000

COMPRESSION



R-2-40 538°C
 Cycle Number: 1245
 N_f : 3693 cycles
 $\Delta \epsilon$: 1.4%



R-2-35 538°C
 Cycle Number: 130
 N_f : 245 cycles
 $\Delta \epsilon$: 5.0%

Figure 237

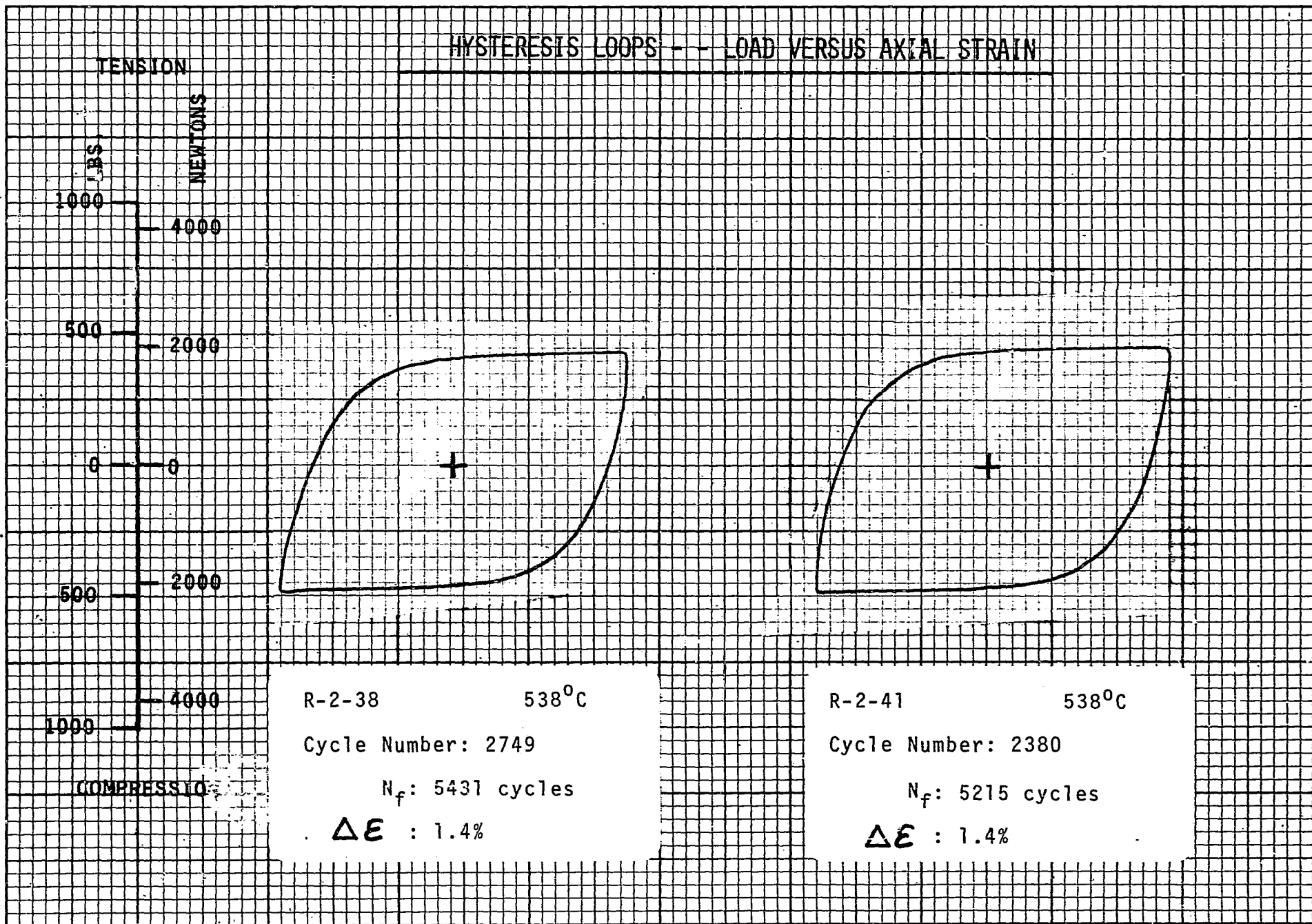


Figure 238

HYSTERESIS LOOPS - LOAD VERSUS AXIAL STRAIN

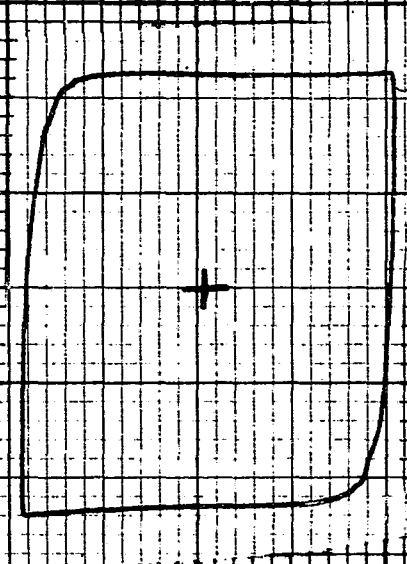
TENSION

1000
500
0
500
1000

NEWTONS

4000
2000
0
2000
4000

COMPRESSION

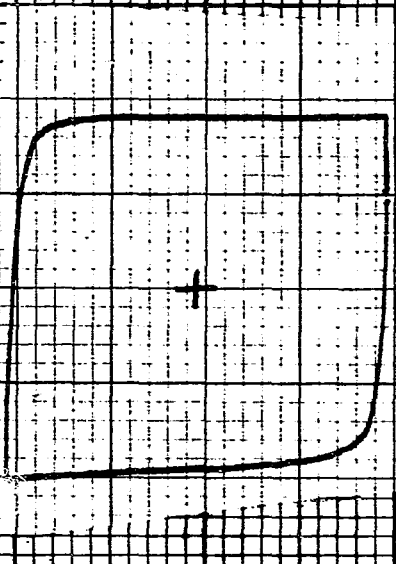


R-2-42 538°C

Cycle Number: 240

N_f : 462 cycles

ΔE : 5.0%



R-2-43 538°C

(200 sec. hold in tension)

Cycle Number: 114

N_f : 211 cycles

ΔE : 5.0%

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

NEWTONS

1000

500

0

500

1000

COMPRESSION

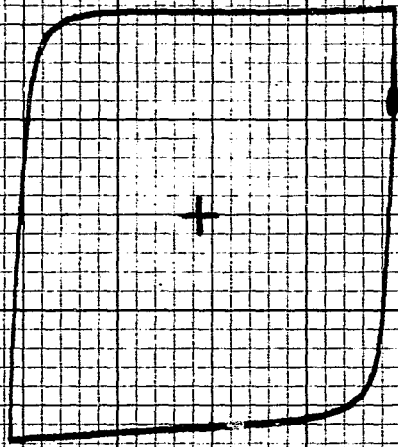
4000

2000

0

2000

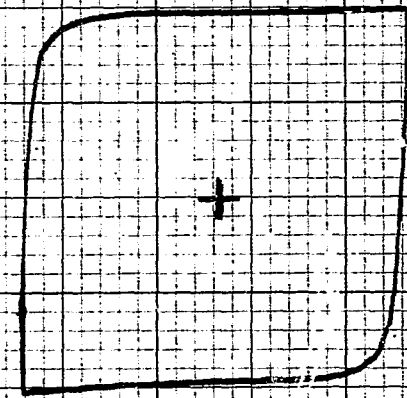
4000



R-2-44 538°C
 (200 sec. hold in tension)
 Cycle Number: 80

N_f : 190 cycles

$\Delta \epsilon$: 5.0%



R-2-45 538°C
 (200 sec. hold in compr.)
 Cycle Number: 65

N_f : 253 cycles

$\Delta \epsilon$: 5.0%

Figure 240

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION

1000 LBS

500

0

500

1000

NEWTONS

4000

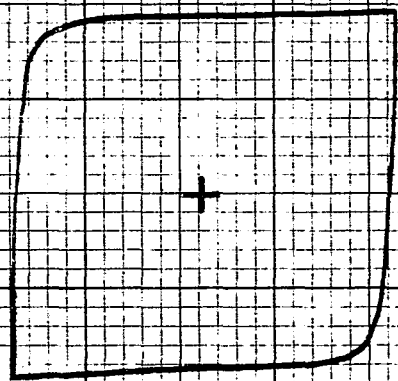
2000

0

2000

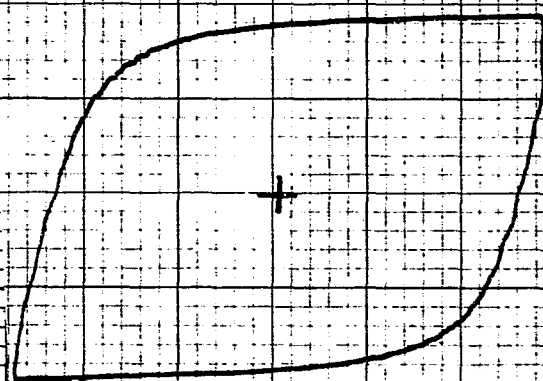
4000

COMPRESSION



R-2-46 538°C
 (200 sec. hold in compr.)
 Cycle Number: 89

N_f : 262 cycles
 ΔE : 5.0%



R-2-47 538°C
 (56 sec. hold in tension)
 Cycle Number: 163

N_f : 1152 cycles
 ΔE : 1.4%

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

LBS

NEWTONS

1000

500

0

500

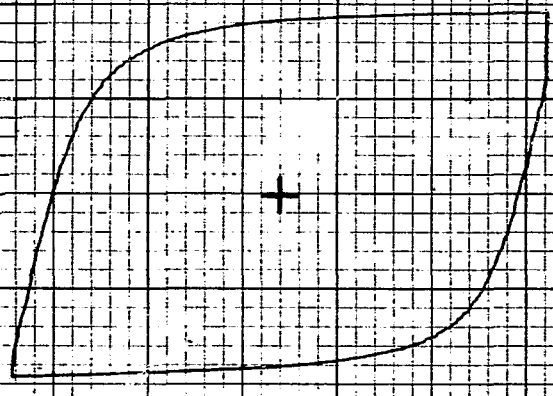
1000

4000

2000

2000

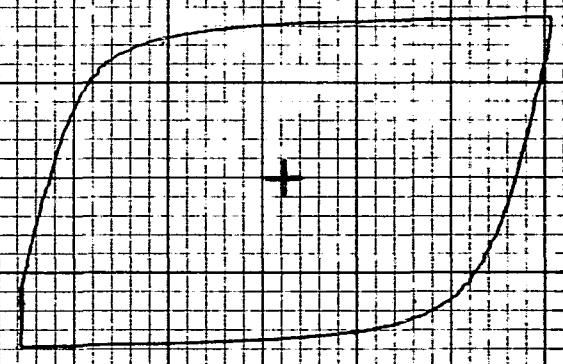
4000



R-2-48 538°C
(56 sec. hold in tension)
Cycle Number: 171

N_f : 1062 cycles

$\Delta \epsilon$: 1.4%



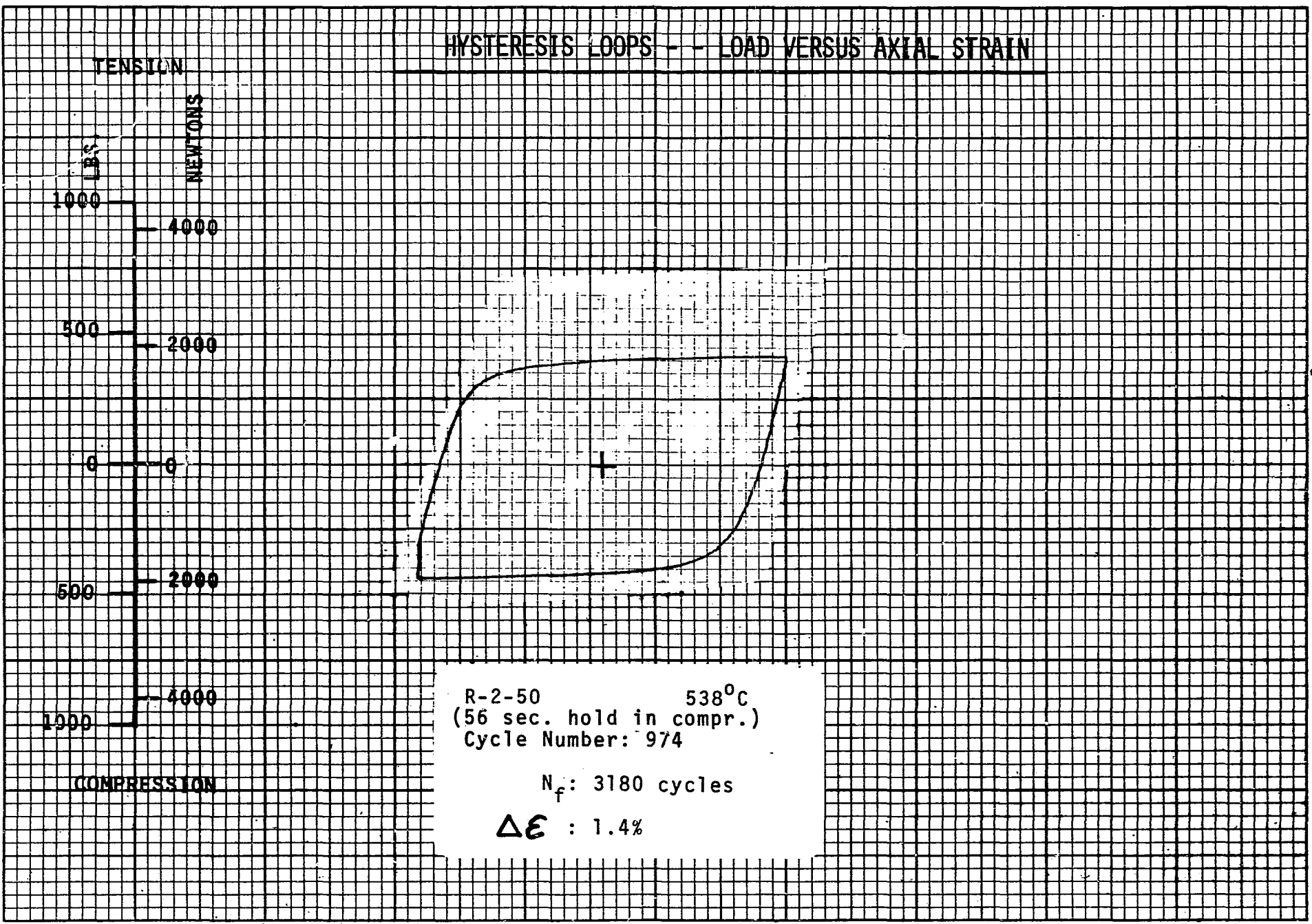
R-2-49 538°C
(56 sec. hold in compr.)
Cycle Number: 264

N_f : 1947 cycles

$\Delta \epsilon$: 1.4%

COMPRESSION

Figure 242



HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

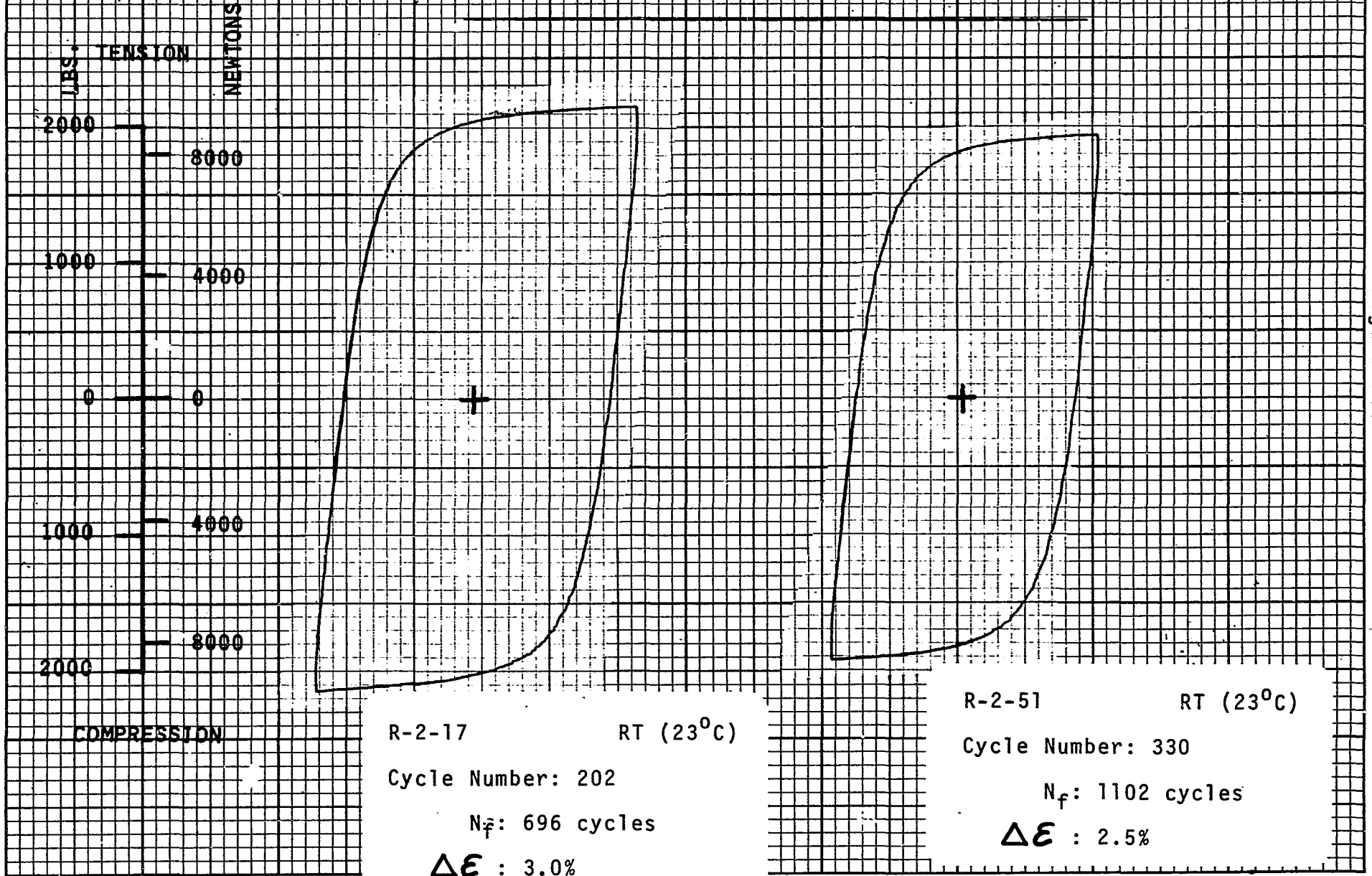


Figure 244

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

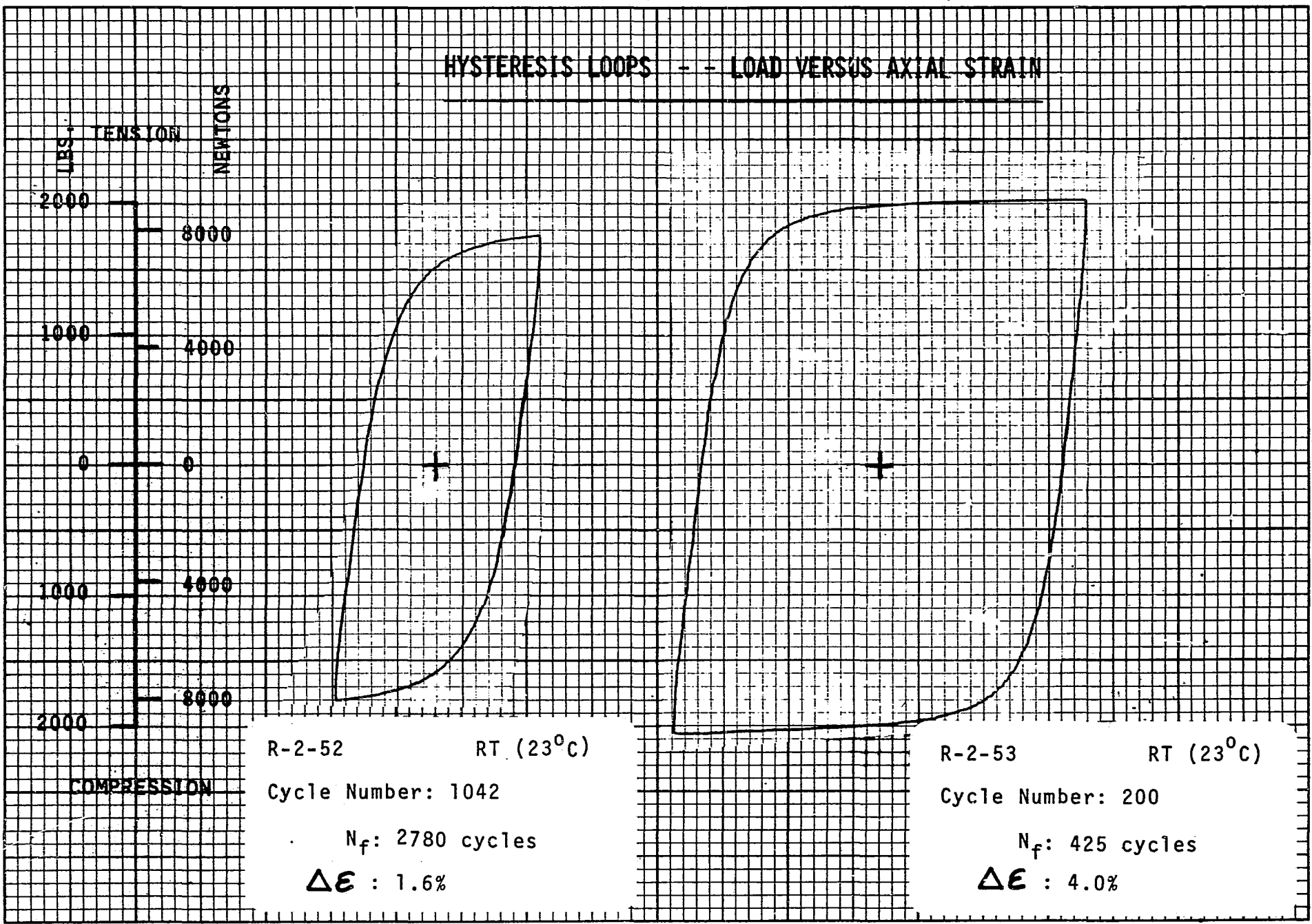
TENSION
LBS.
NEWTONS

2000
1000
0
1000
2000

R-2-52 RT (23°C)
Cycle Number: 1042
 N_f : 2780 cycles
 $\Delta\epsilon$: 1.6%

R-2-53 RT (23°C)
Cycle Number: 200
 N_f : 425 cycles
 $\Delta\epsilon$: 4.0%

COMPRESSION



HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

LBS. TENSION
NEWTONS

2000
1000
0
1000
2000

8000
4000
0
4000
8000

COMPRESSION

R-2-54 RT(23°C)

Cycle Number 1600

N_f : 3283 cycles

ΔE : 1.4%

R-2-55

RT (23°C)

Cycle Number: 69

N_f : 141 cycles

ΔE : 3.0%

Figure 246

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

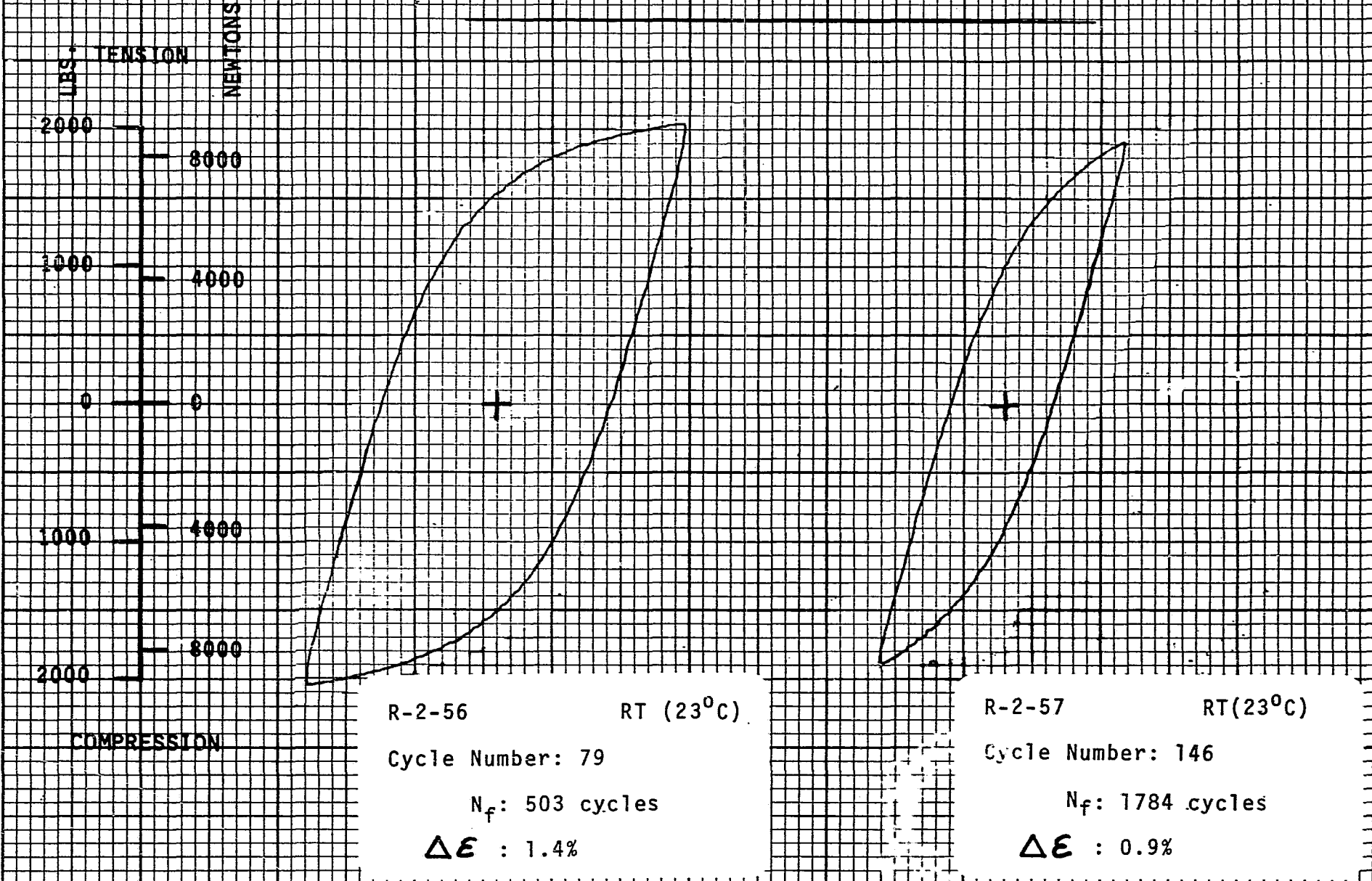


Figure 247

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

LBS: TENSION
NEWTONS

2000

8000

1000

4000

0

0

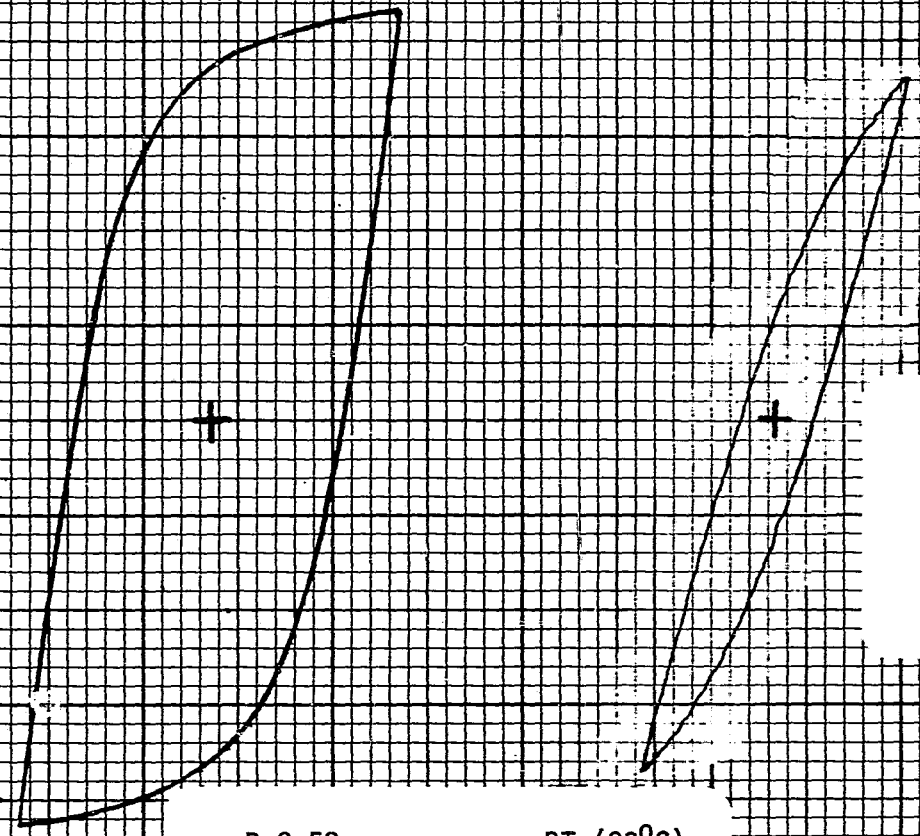
1000

4000

2000

8000

COMPRESSION



R-2-58 RT (23°C)

Cycle Number: 50

N_f : 279 cycles

$\Delta \epsilon$: 2.0%

R-2-59

RT (23°C)

Cycle Number: 240

N_f : 5607 cycles

$\Delta \epsilon$: 0.7%

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

LBS.
TENSION
NEWTONS

2000

8000

1000

4000

0

0

1000

4000

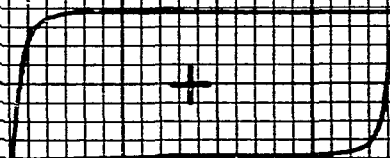
2000

8000

COMPRESSION

R-20-1 538^oC

Cycle Number: 113

N_f: 390 cycles $\Delta \epsilon$: 5.0%

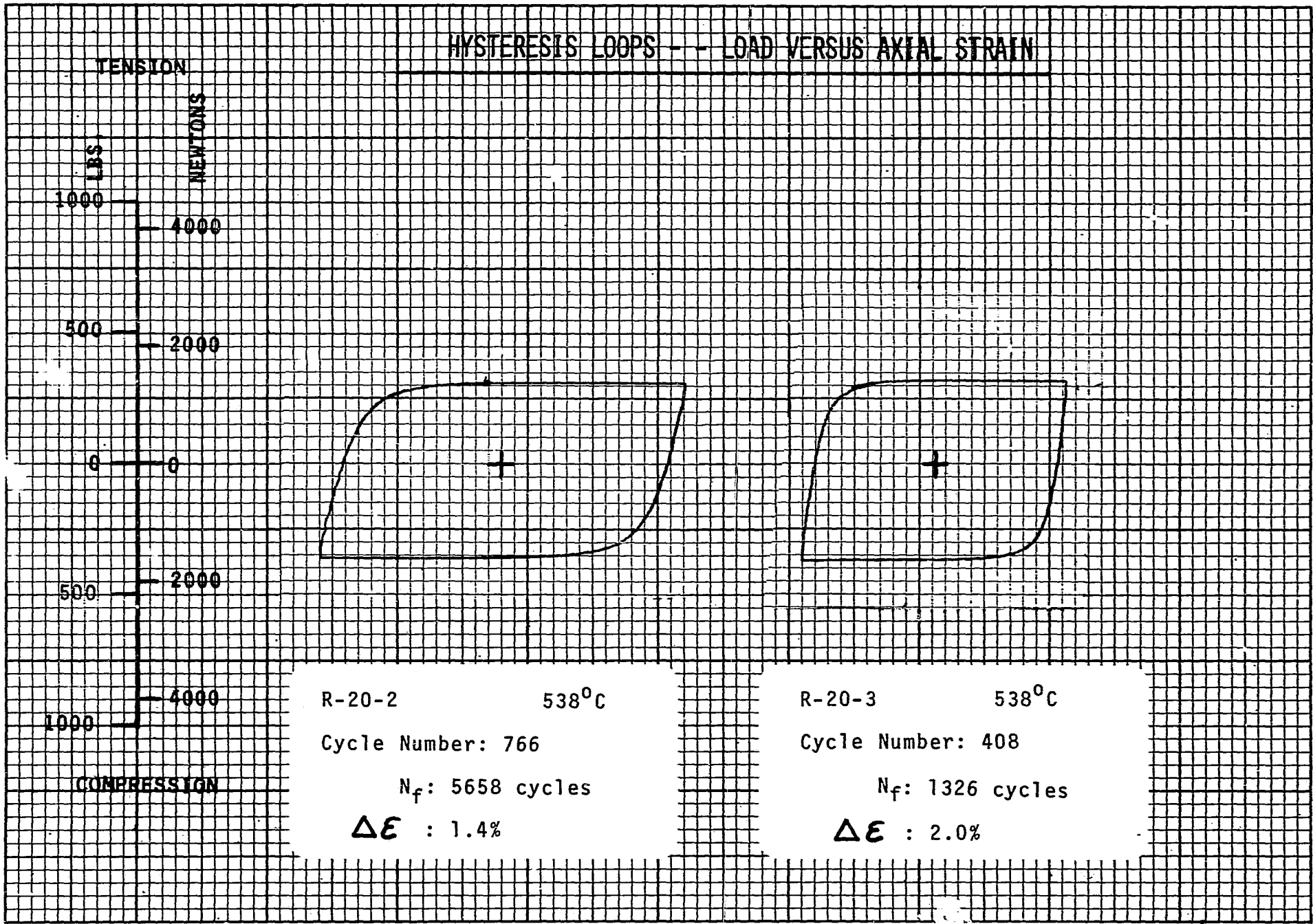
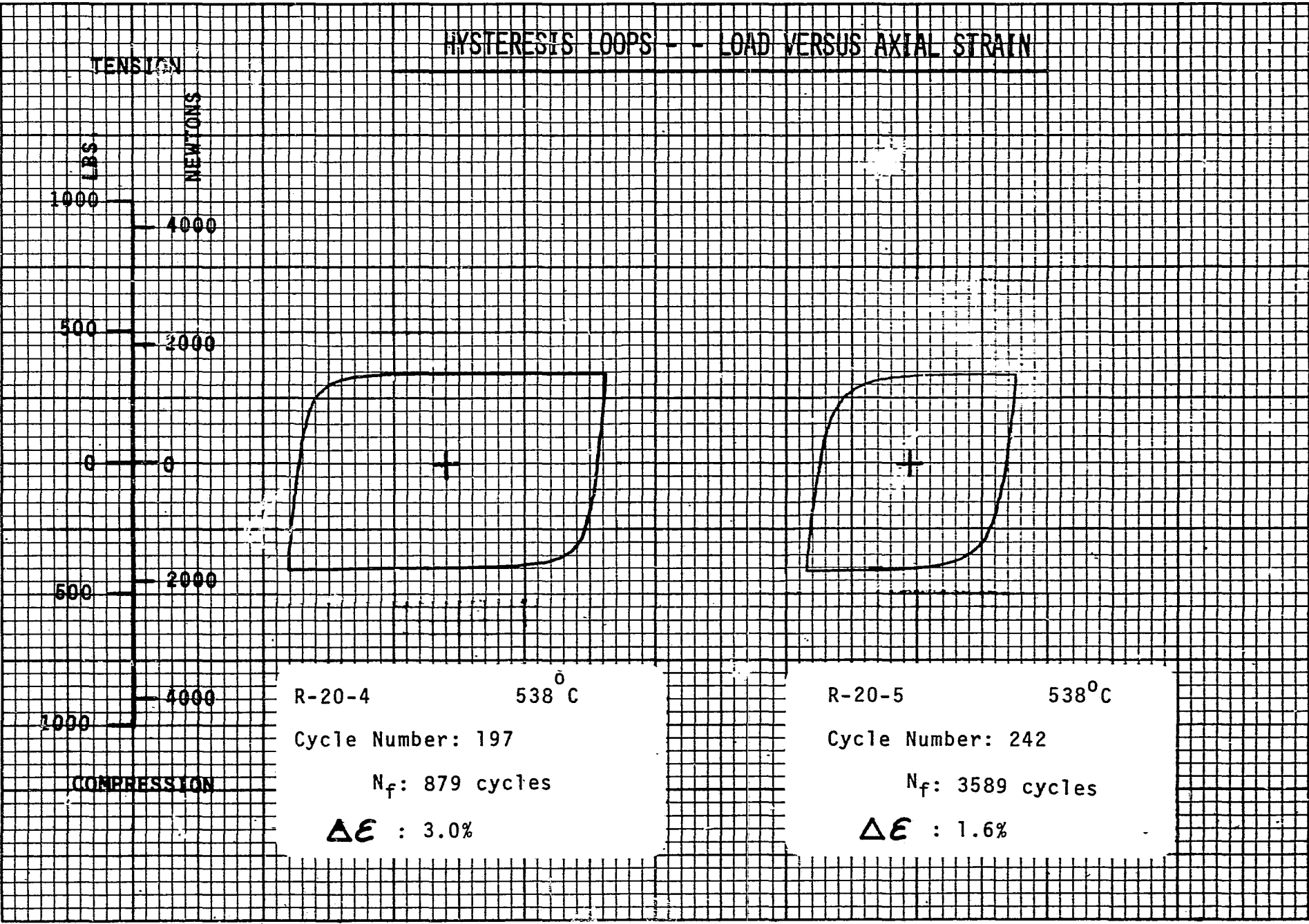


Figure 250

Figure 251



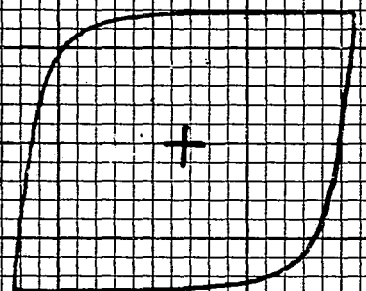
HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

1000
500
0
500
1000

NEWTONS

4000
2000
0
2000
4000

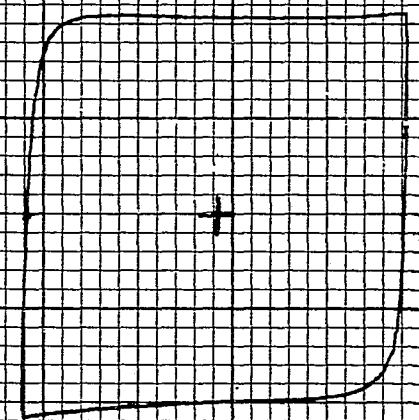


R-20-6 538°C

Cycle Number: 218

N_f : 2700 cycles

ΔE : 1.8%



R-2-67 538°C

(300 sec. hold in tension)

Cycle Number: 23

N_f : 218 cycles

ΔE : 5.0%

COMPRESSION

Figure 252

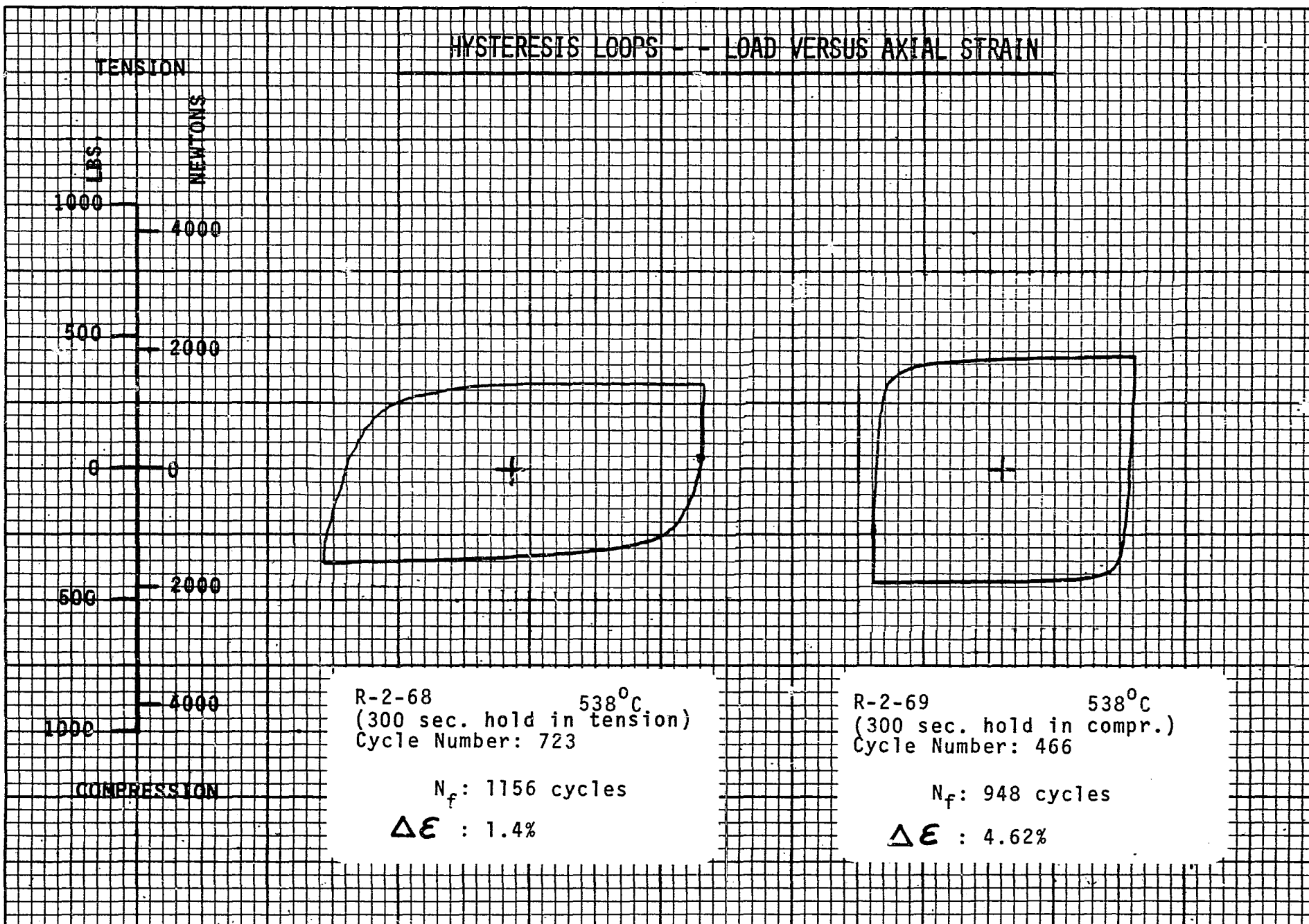


Figure 253

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

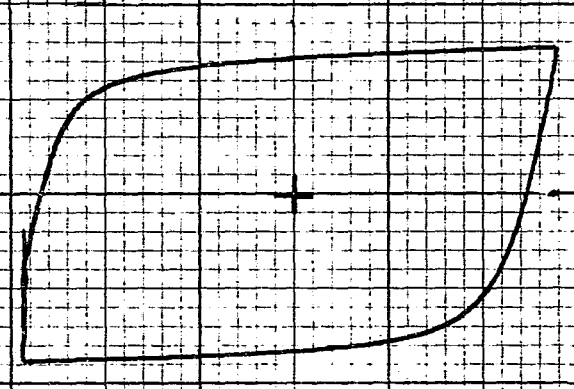
TENSION

1000
500
0
500
1000

NEWTONS

4000
2000
0
2000
4000

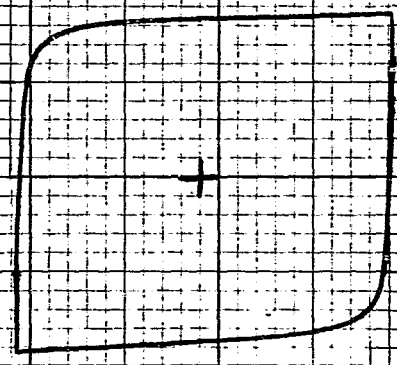
COMPRESSION



R-2-72 538°C
 (300 sec. hold in compr.)
 Cycle Number: 256

N_f : 1224 cycles

$\Delta \epsilon$: 1.4%



R-2-73 538°C
 (300 sec. hold in tension
 and compression)
 Cycle Number: 117

N_f : 234 cycles

$\Delta \epsilon$: 5.0%

Figure 254

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION
LBS
NEWTONS

1000

4000

500

2000

0

0

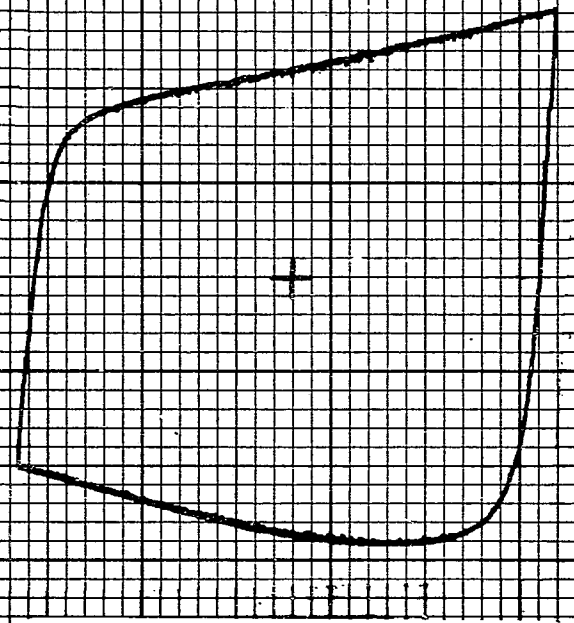
500

2000

1000

4000

COMPRESSION

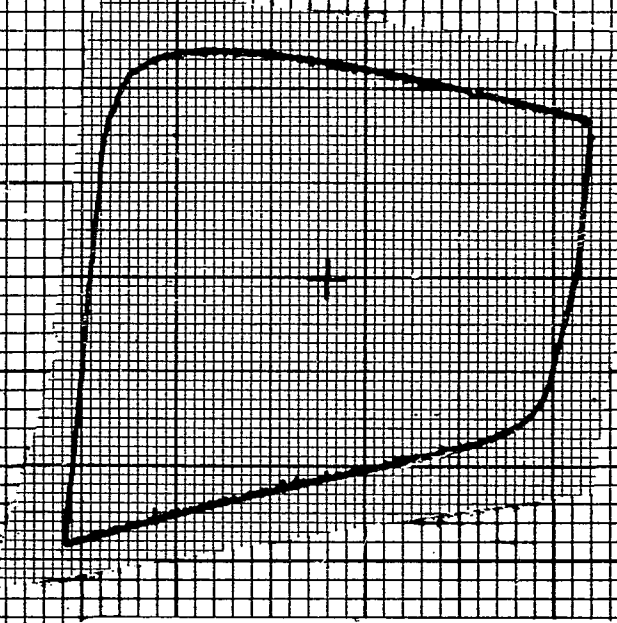


R-2-61 538⁰C/260⁰C

Cycle Number: 198

N_f: 284 cycles

$\Delta\epsilon$: 3.5%



R-2-62 538⁰C/260⁰C

Cycle Number: 119

N_f: 188 cycles

$\Delta\epsilon$: 3.5%

Figure 255

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

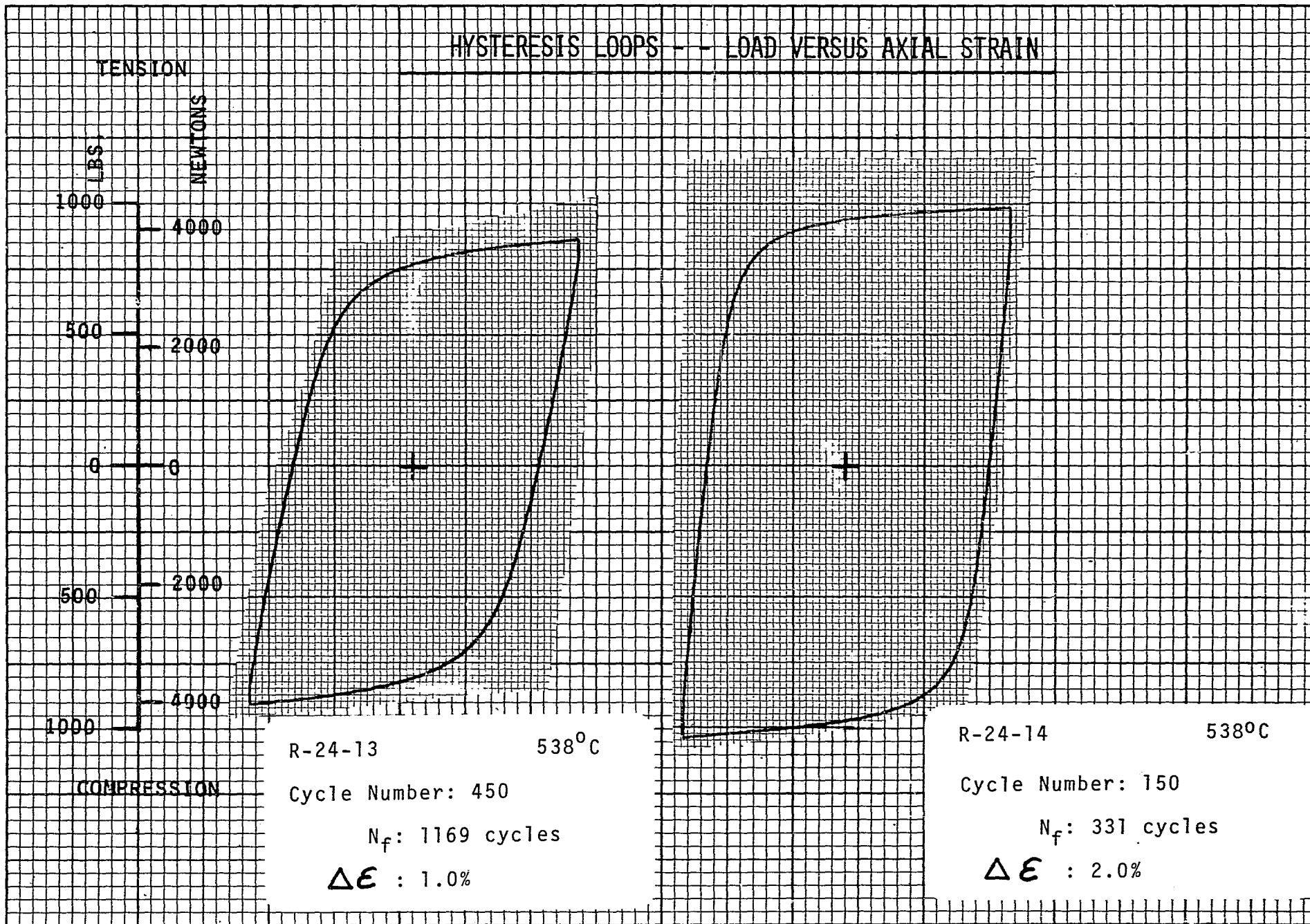
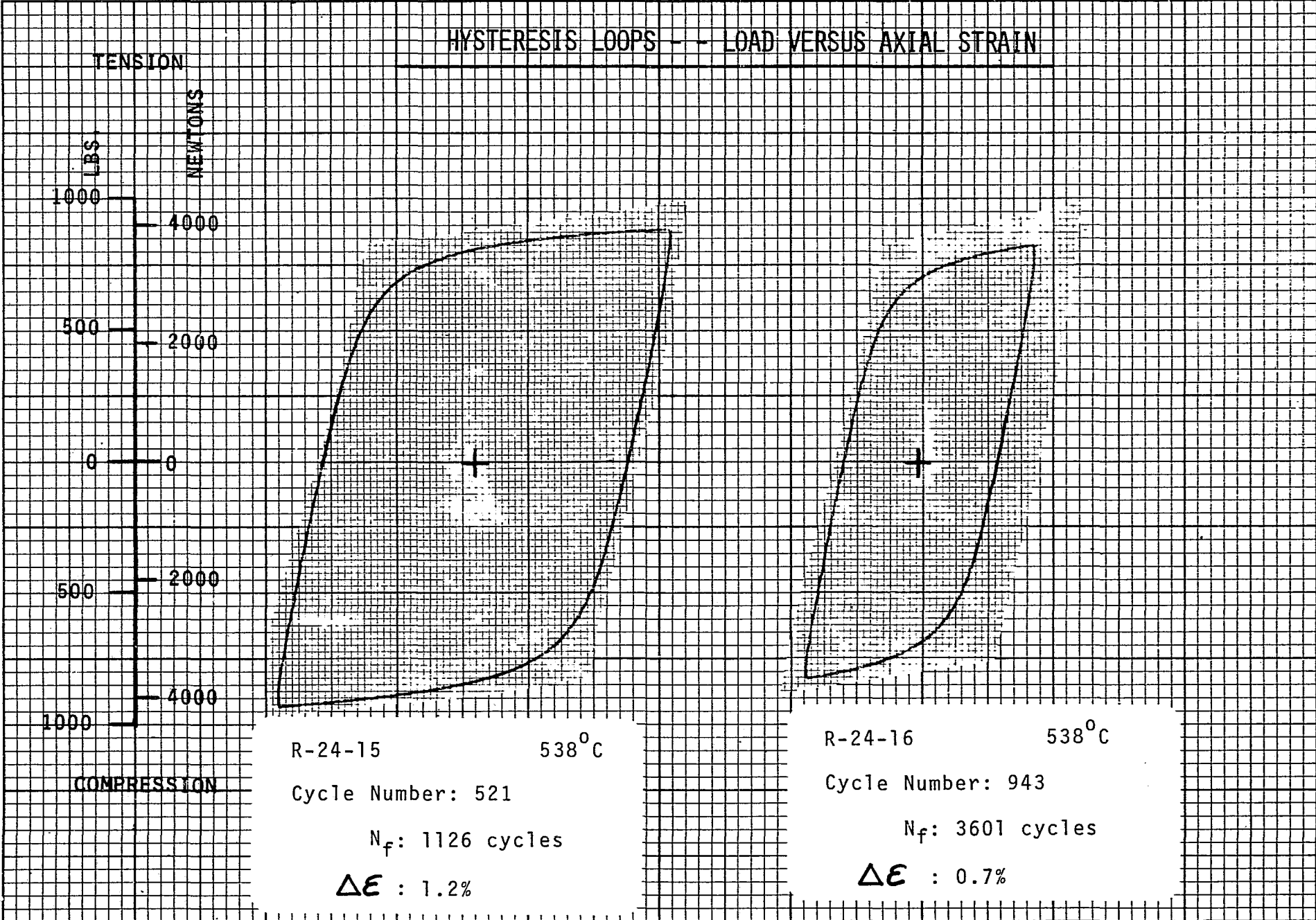


Figure 256



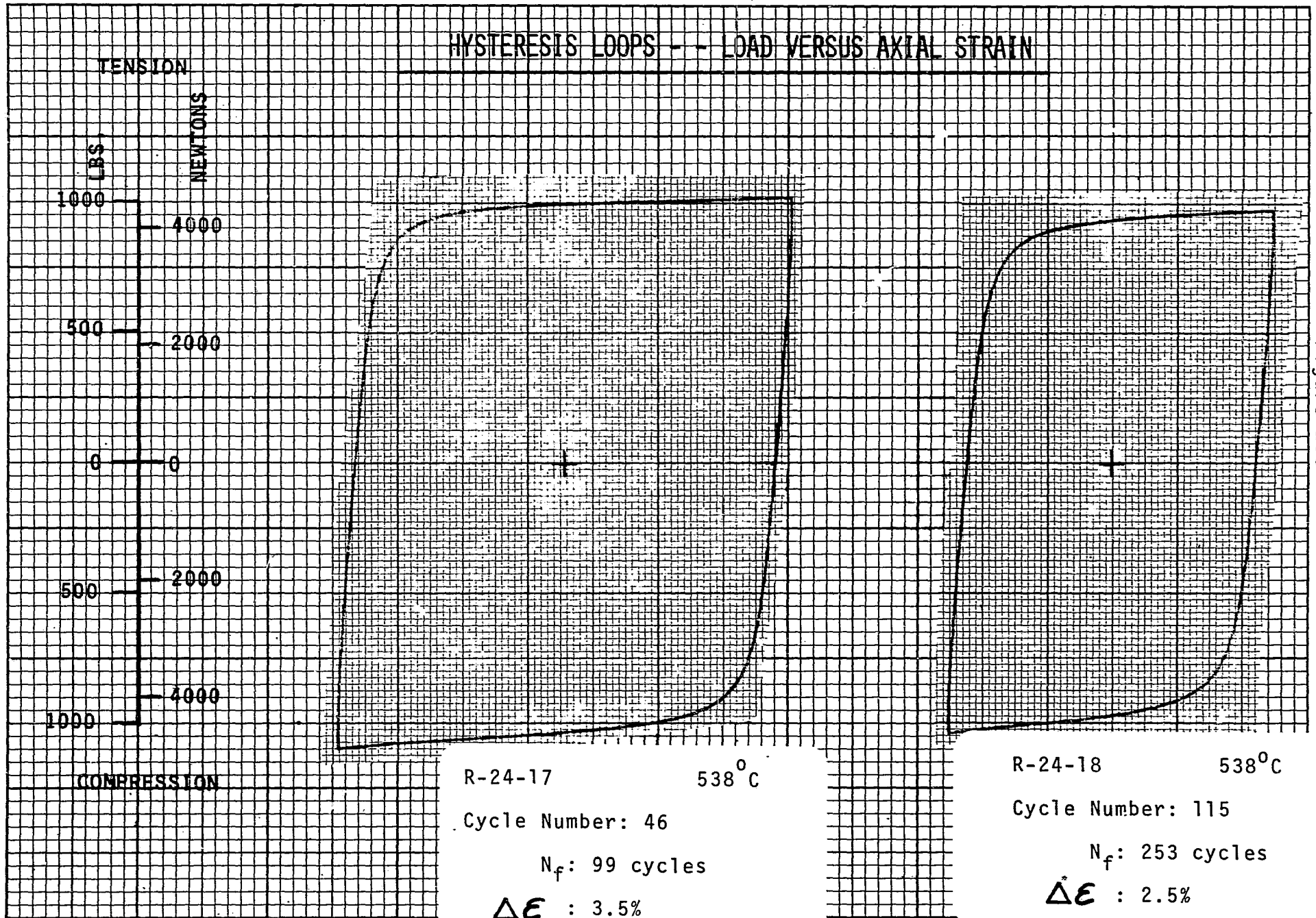
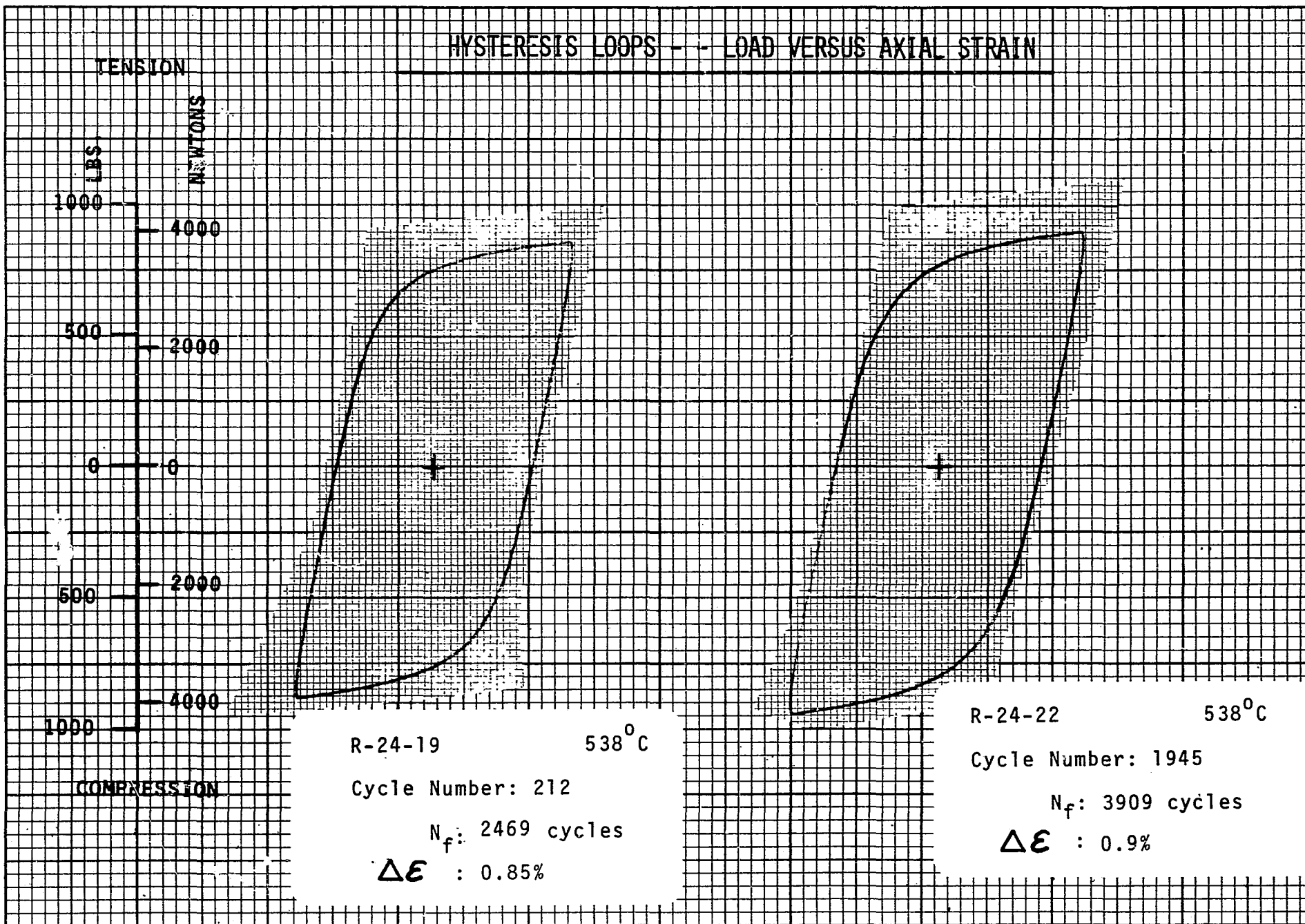


Figure 258



HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

LBS

NEWTONS

1000

4000

500

2000

0

0

500

2000

1000

4000

COMPRESSION

R-24-21 538°C

Cycle Number: 1330

N_f : 3586 cycles

ΔE : 0.9%

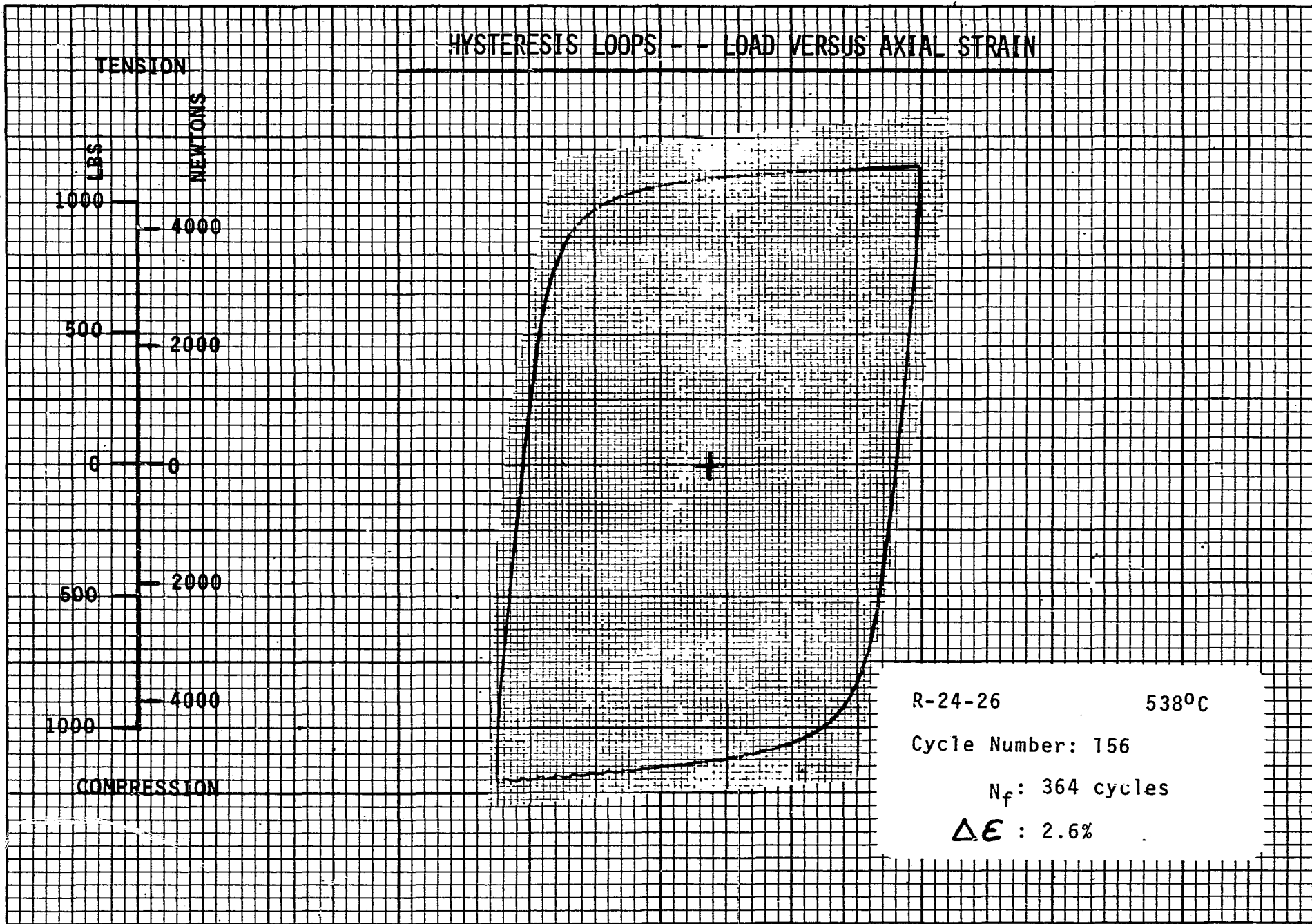
R-24-23 538°C

Cycle Number: 200

N_f : 339 cycles

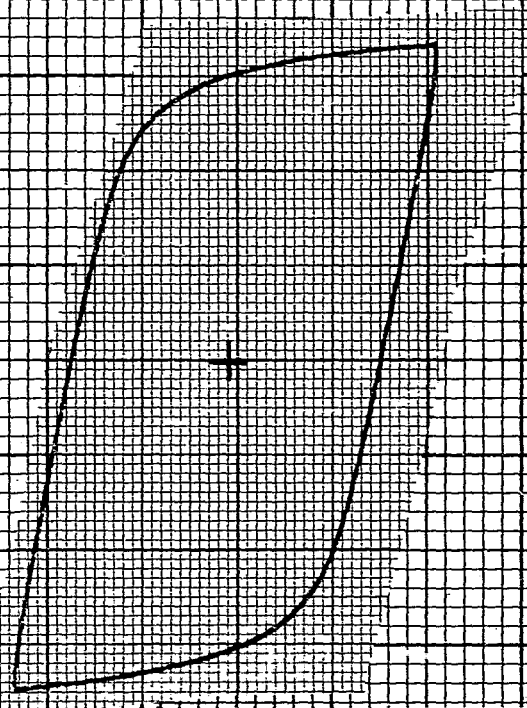
ΔE : 2.6%

Figure 260



HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION
LBS
1000
500
0
500
1000
NEWTONS
4000
2000
0
2000
4000
COMPRESSION

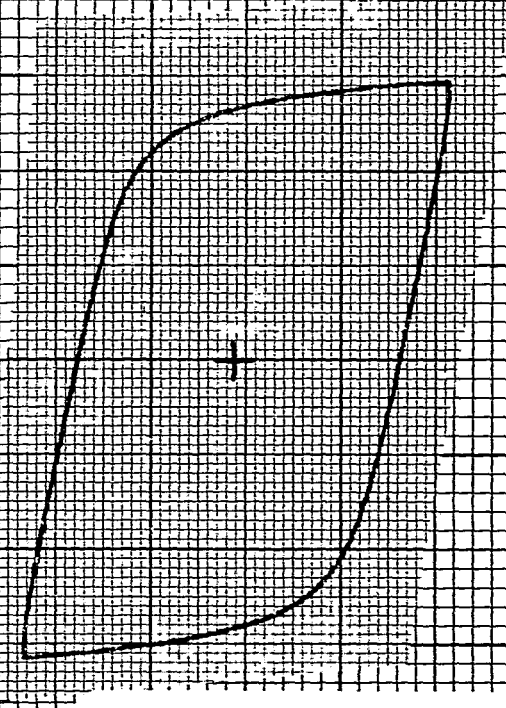


R-24-20 538°C

Cycle Number: 40

N_f : 1138 cycles

$\Delta \epsilon$: 0.9%



R-24-25 538°C

Cycle Number: 314

N_f : 1196 cycles

$\Delta \epsilon$: 0.9%

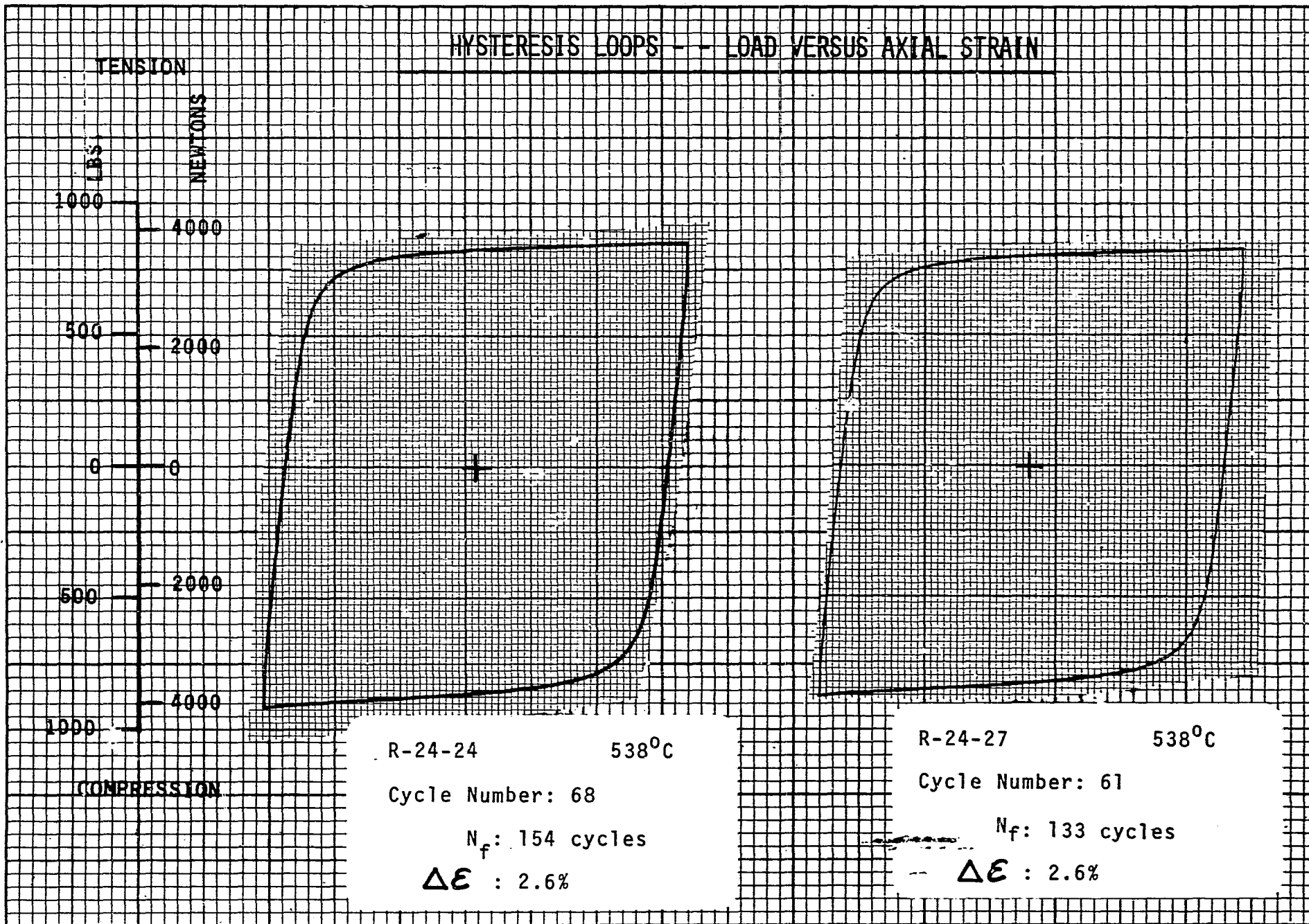
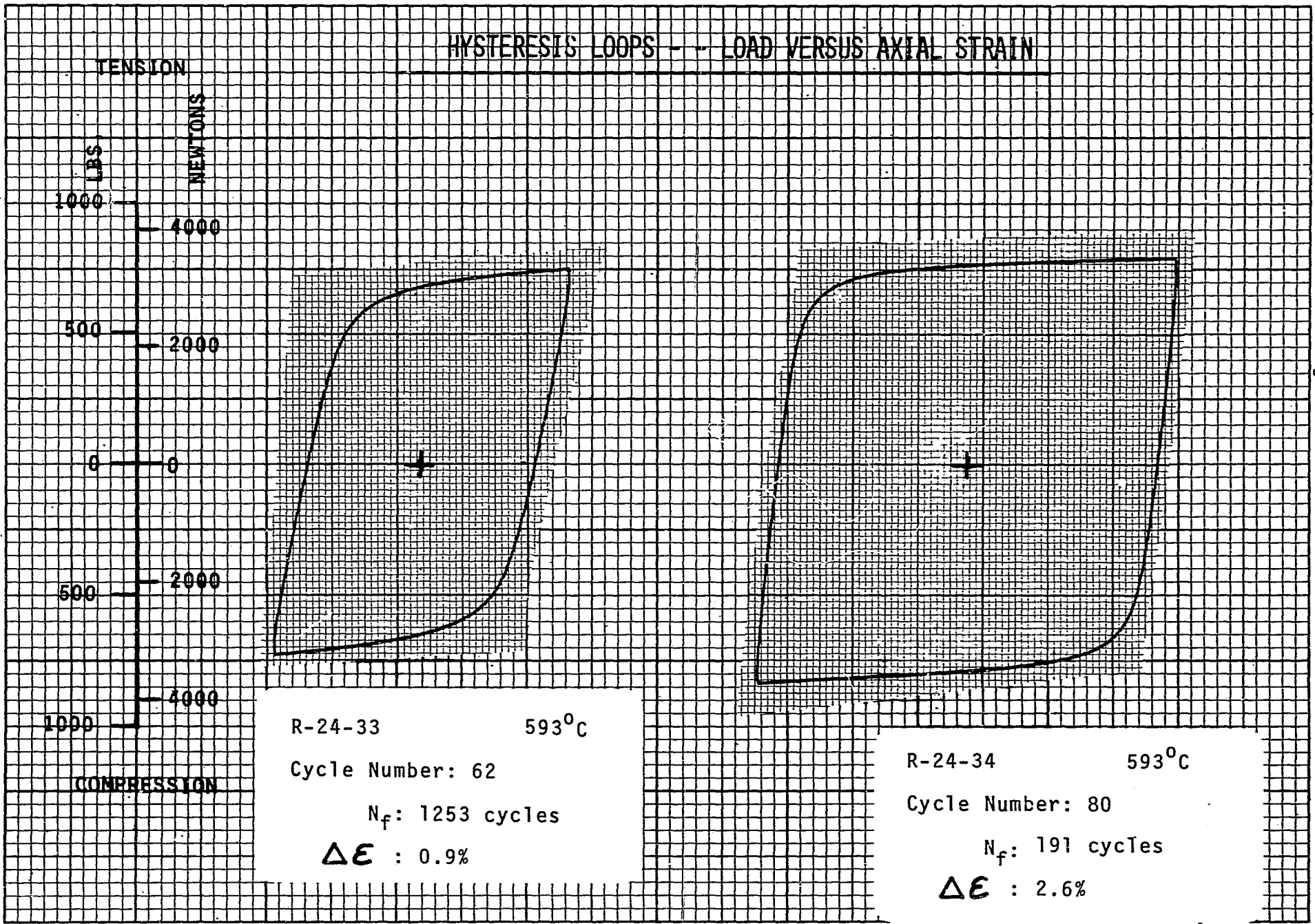


Figure 263

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN



R-24-33 593°C

Cycle Number: 62

N_f : 1253 cycles

ΔE : 0.9%

R-24-34 593°C

Cycle Number: 80

N_f : 191 cycles

ΔE : 2.6%

Figure 264

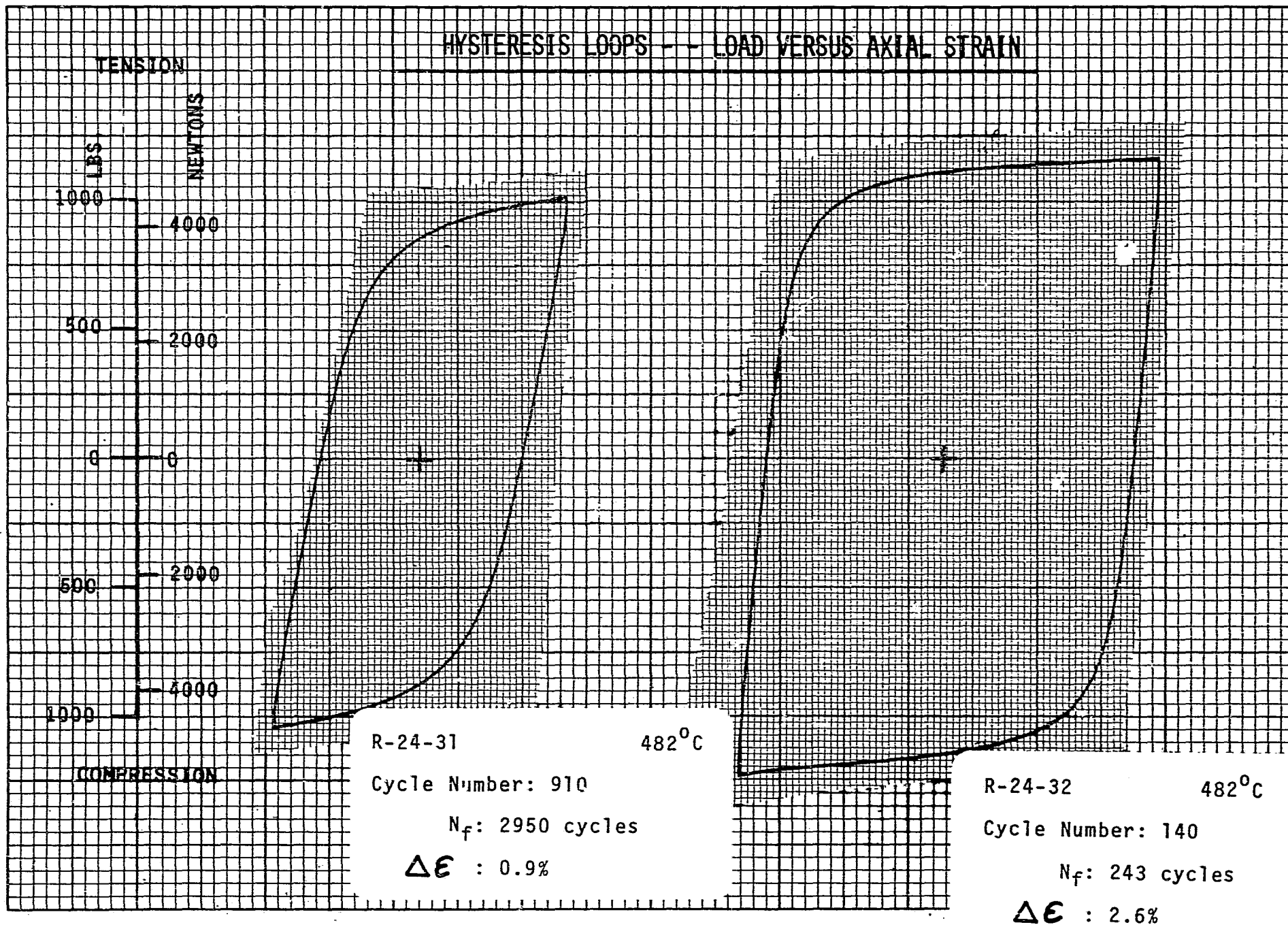


Figure 265

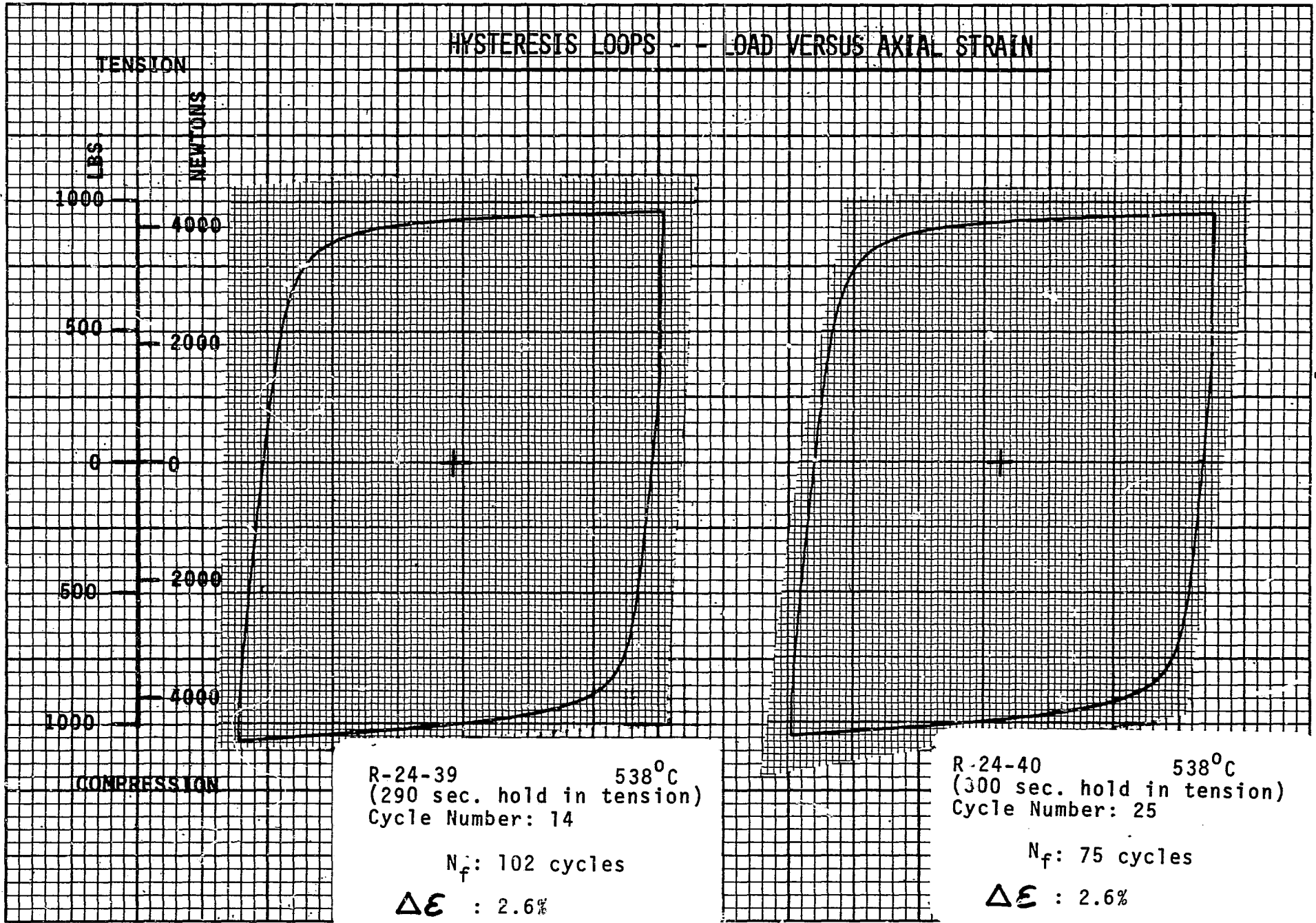
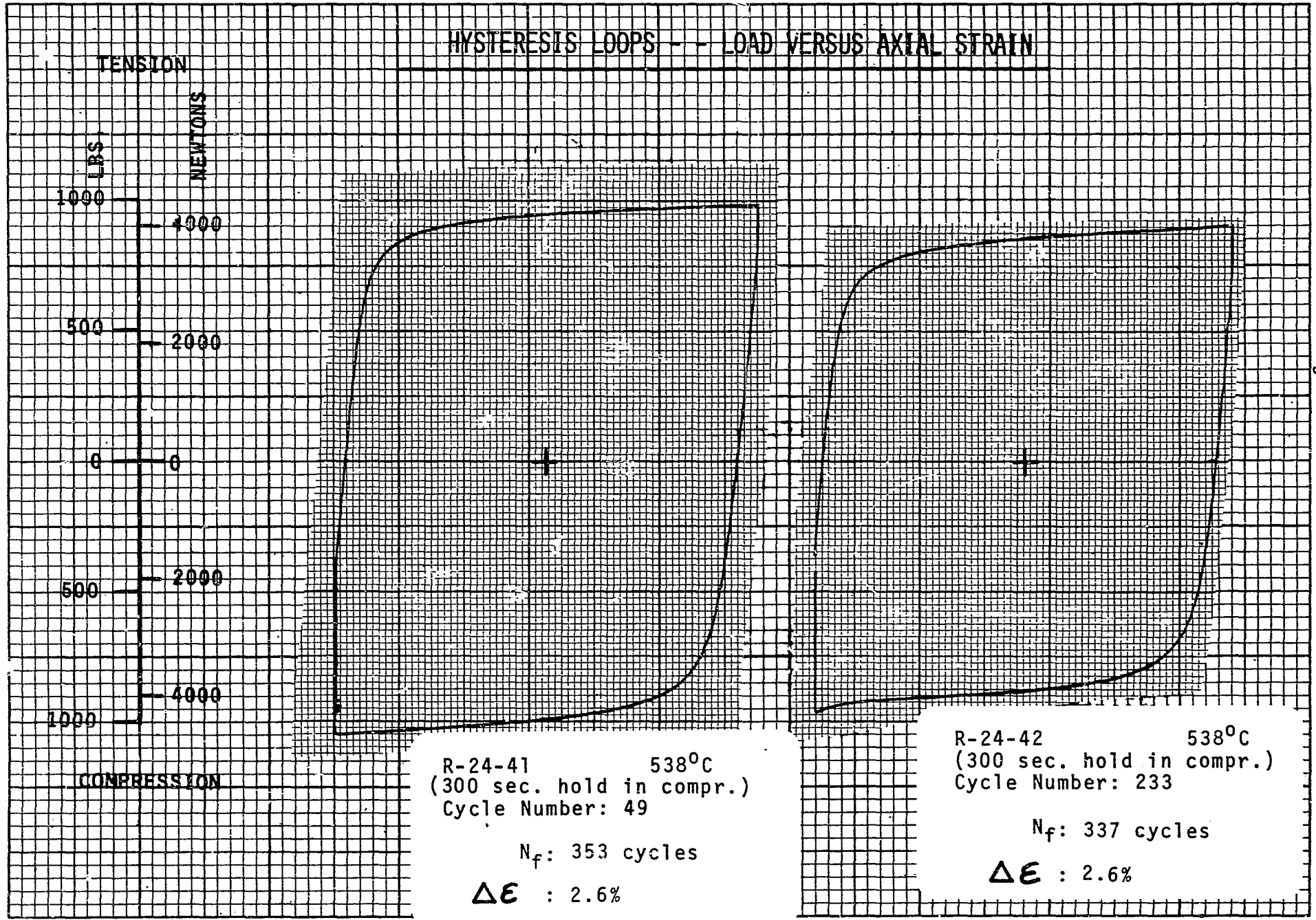


Figure 266

Figure 267



HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

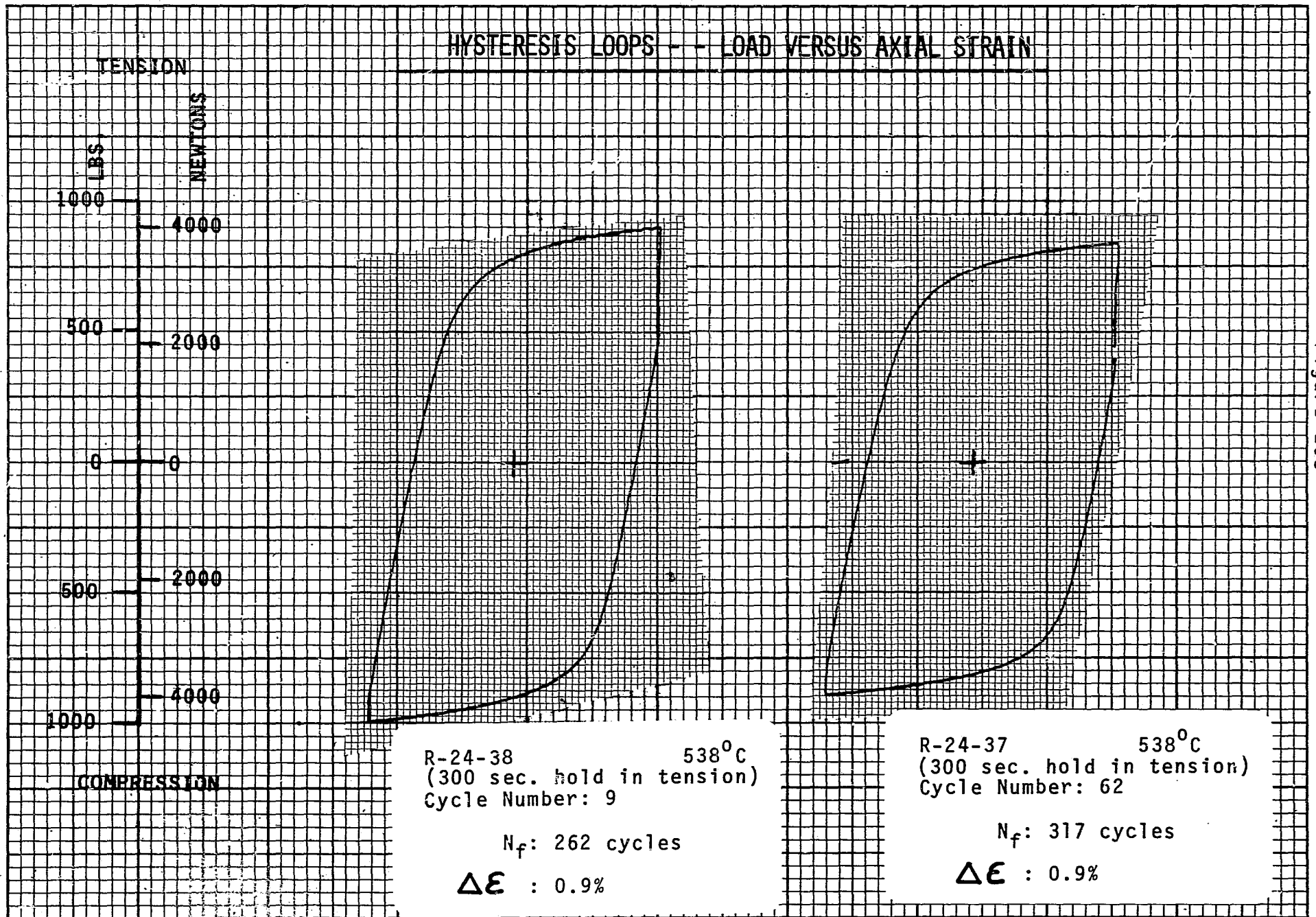


Figure 268

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

LBS

NEWTONS

1000

500

0

500

1000

4000

2000

0

2000

4000

COMPRESSION

R-24-43 538°C
 (300 sec. hold in compr.)
 Cycle Number: 1410

N_f : 2981 cycles

ΔE : 0.9%

R-24-45 538°C
 (300 sec. hold in compr.)
 Cycle Number: 1632

N_f : 3392 cycles

ΔE : 0.9%

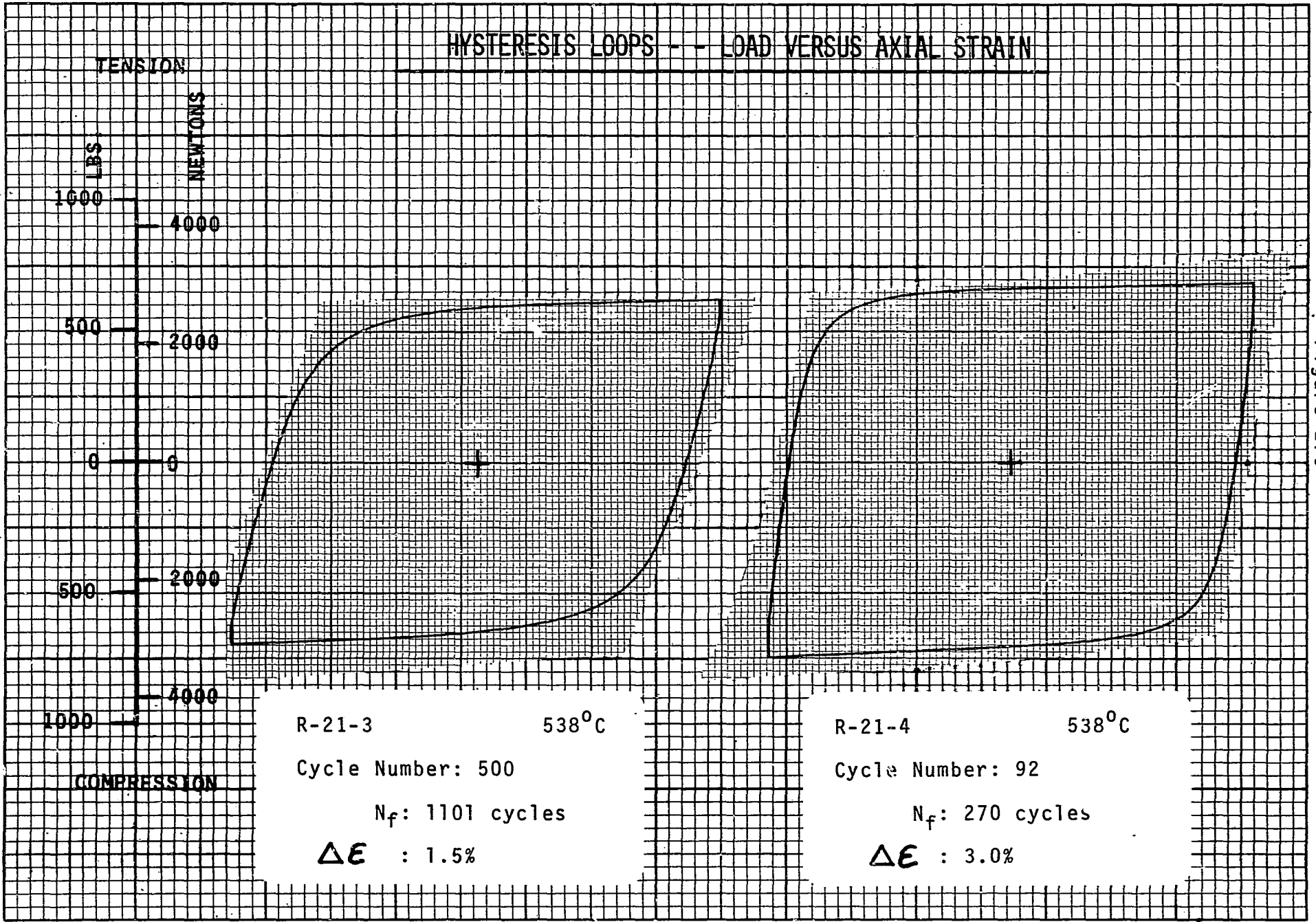
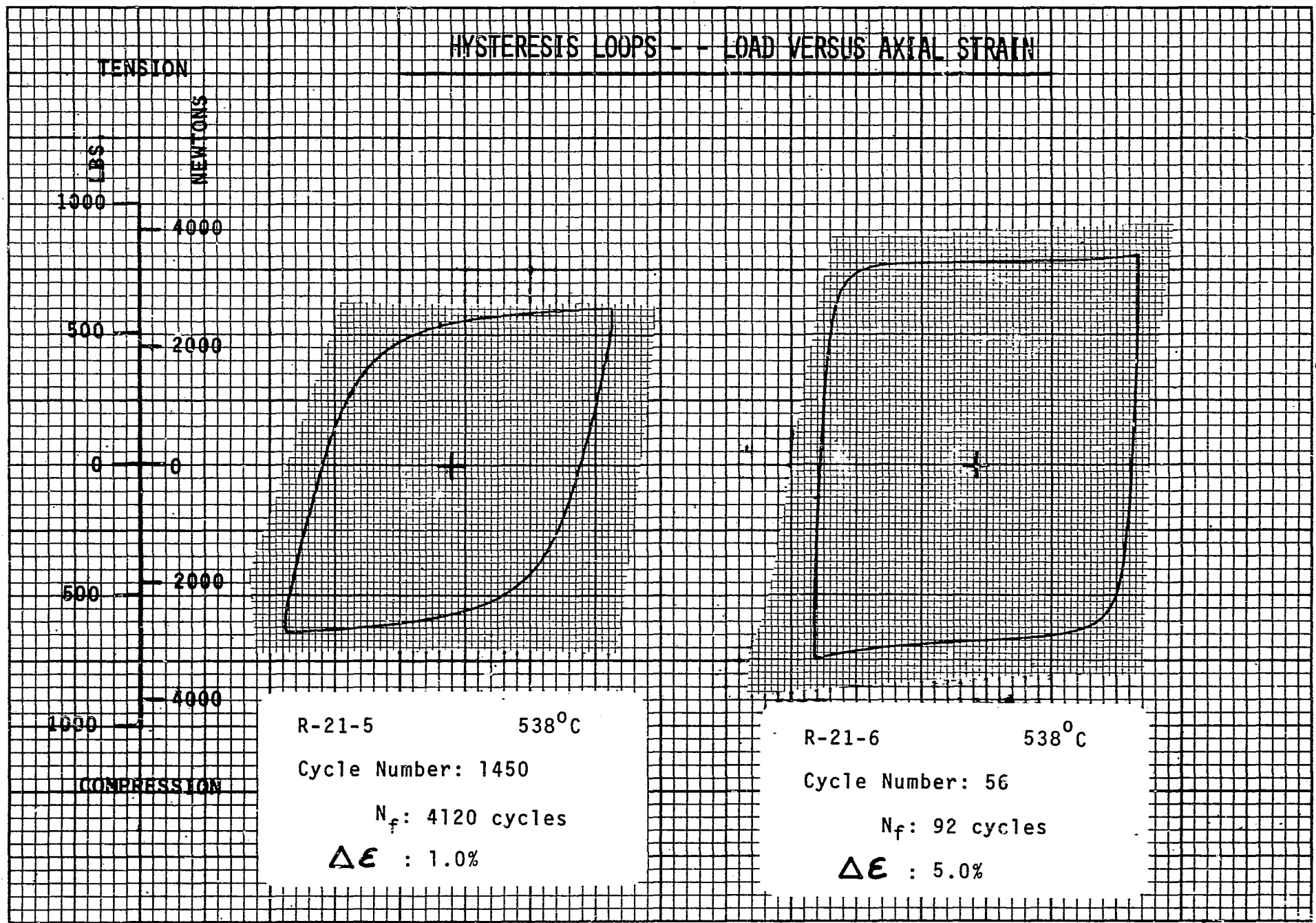


Figure 270

Figure 271



HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

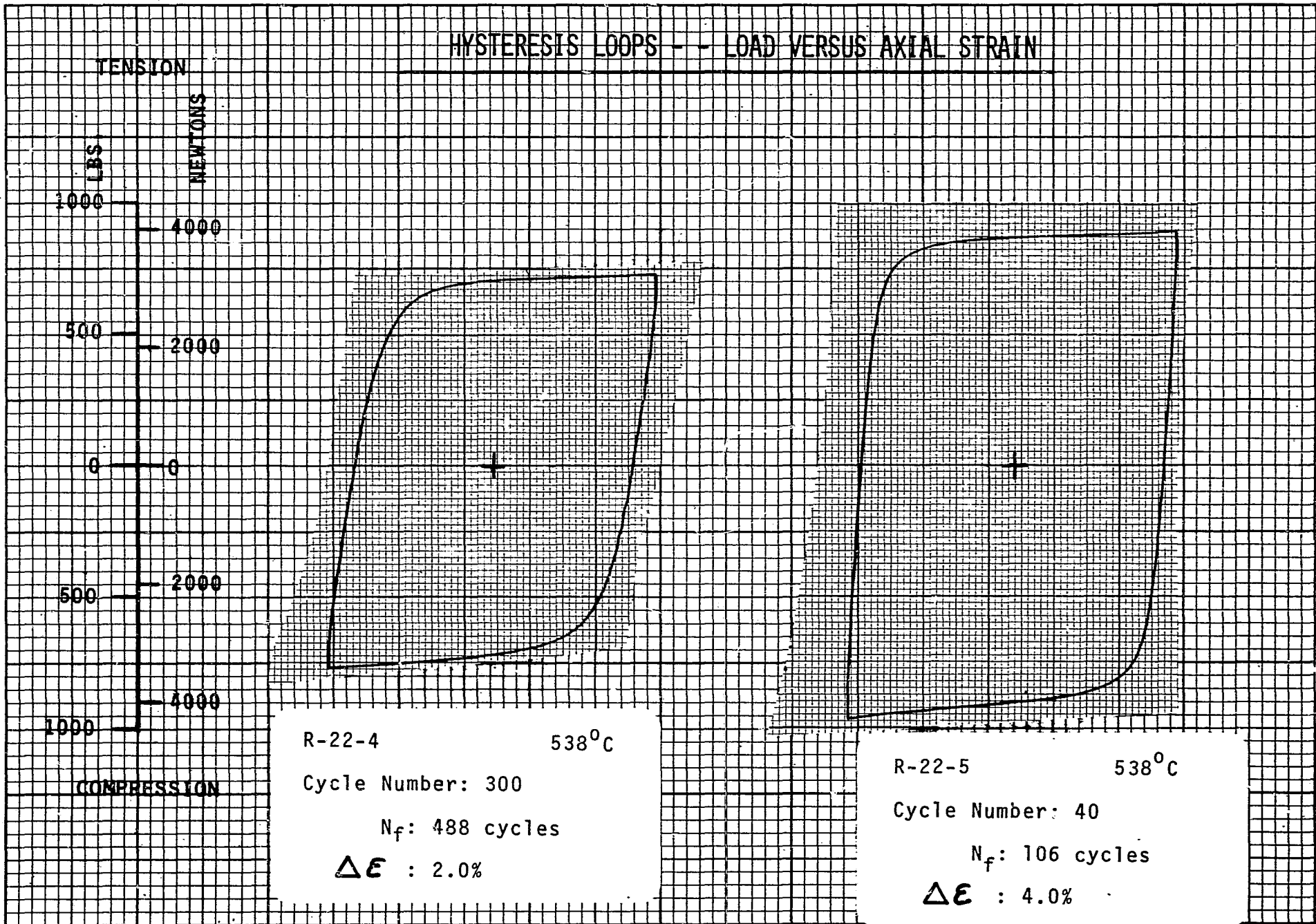


Figure 272

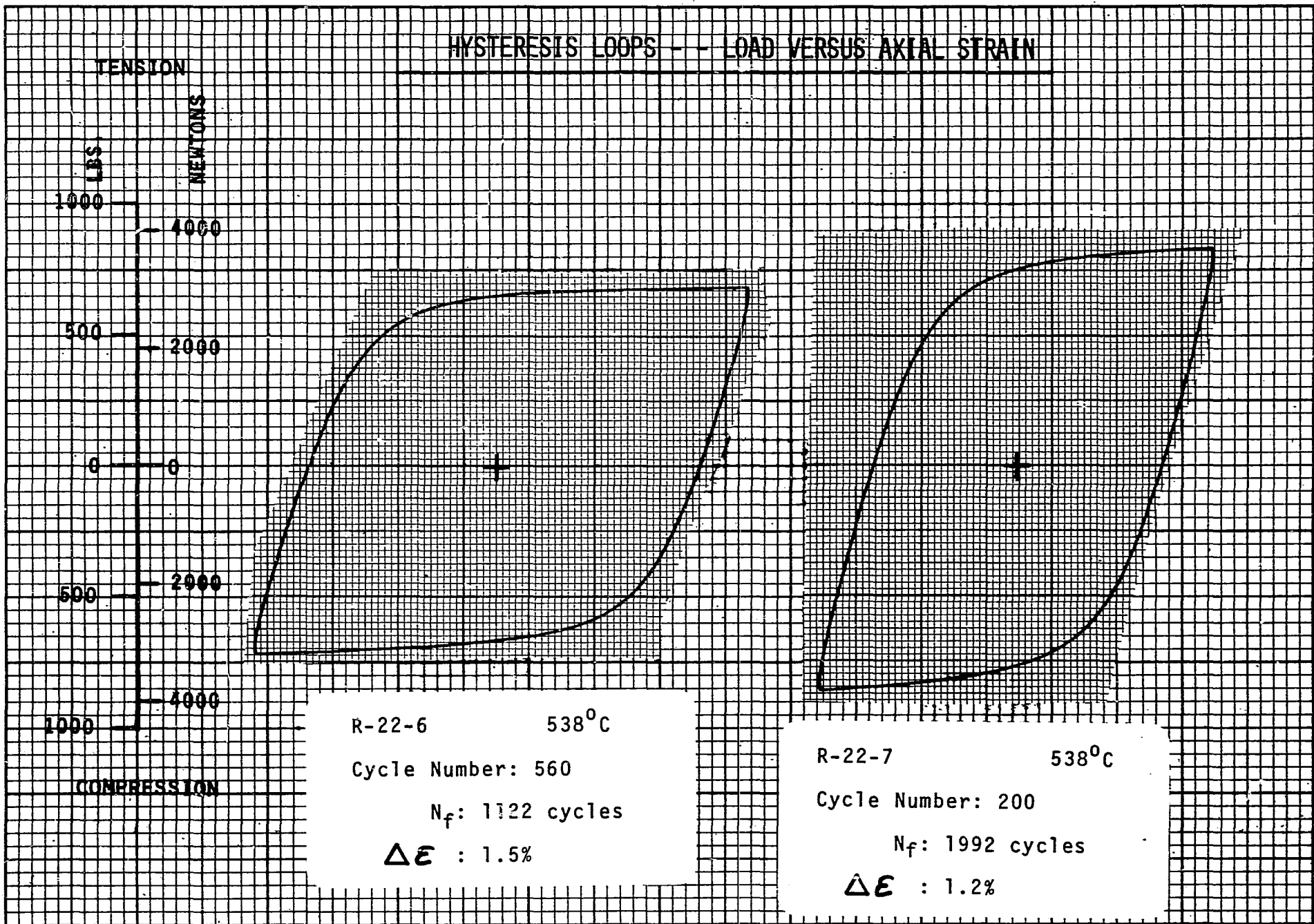


Figure 273

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION

NEWTONS

LBS

1000

500

0

500

1000

4000

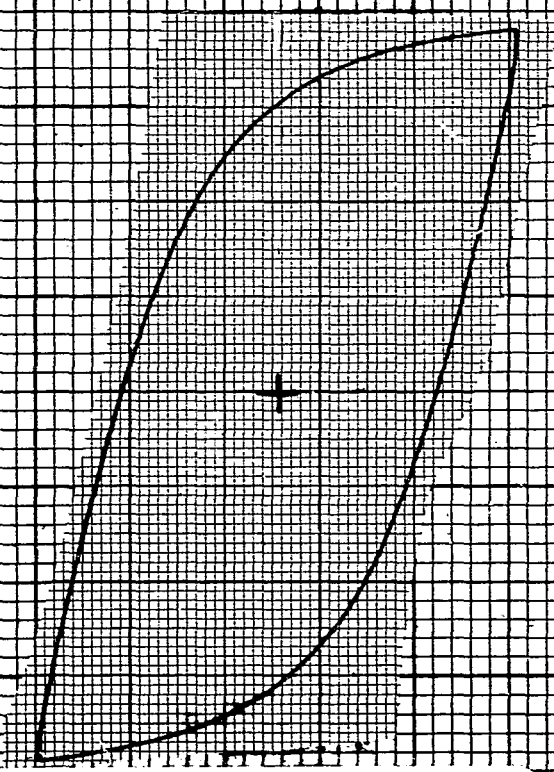
2000

0

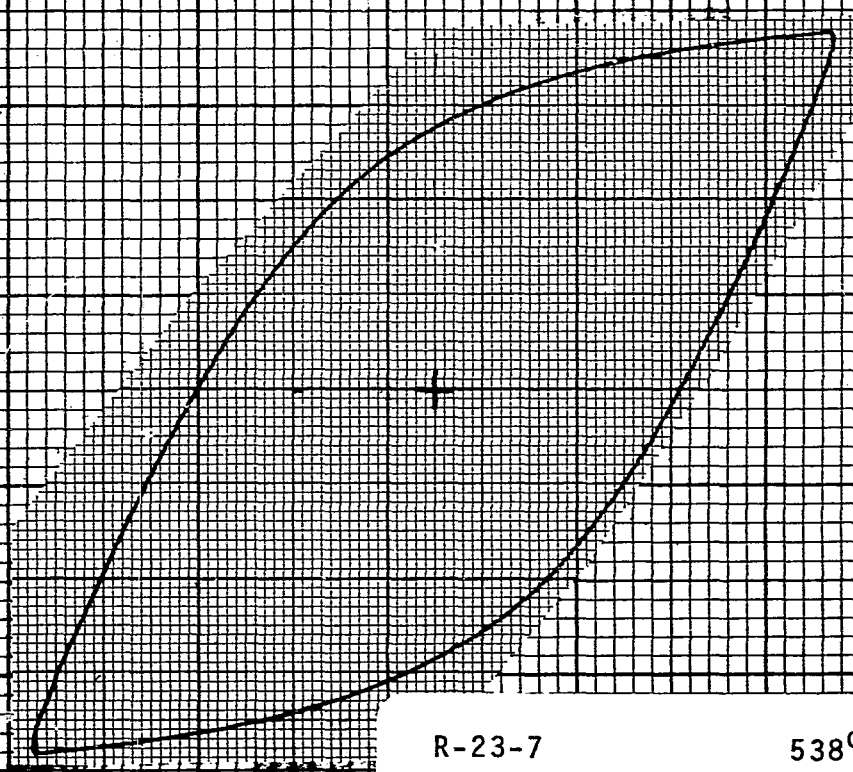
2000

4000

COMPRESSION

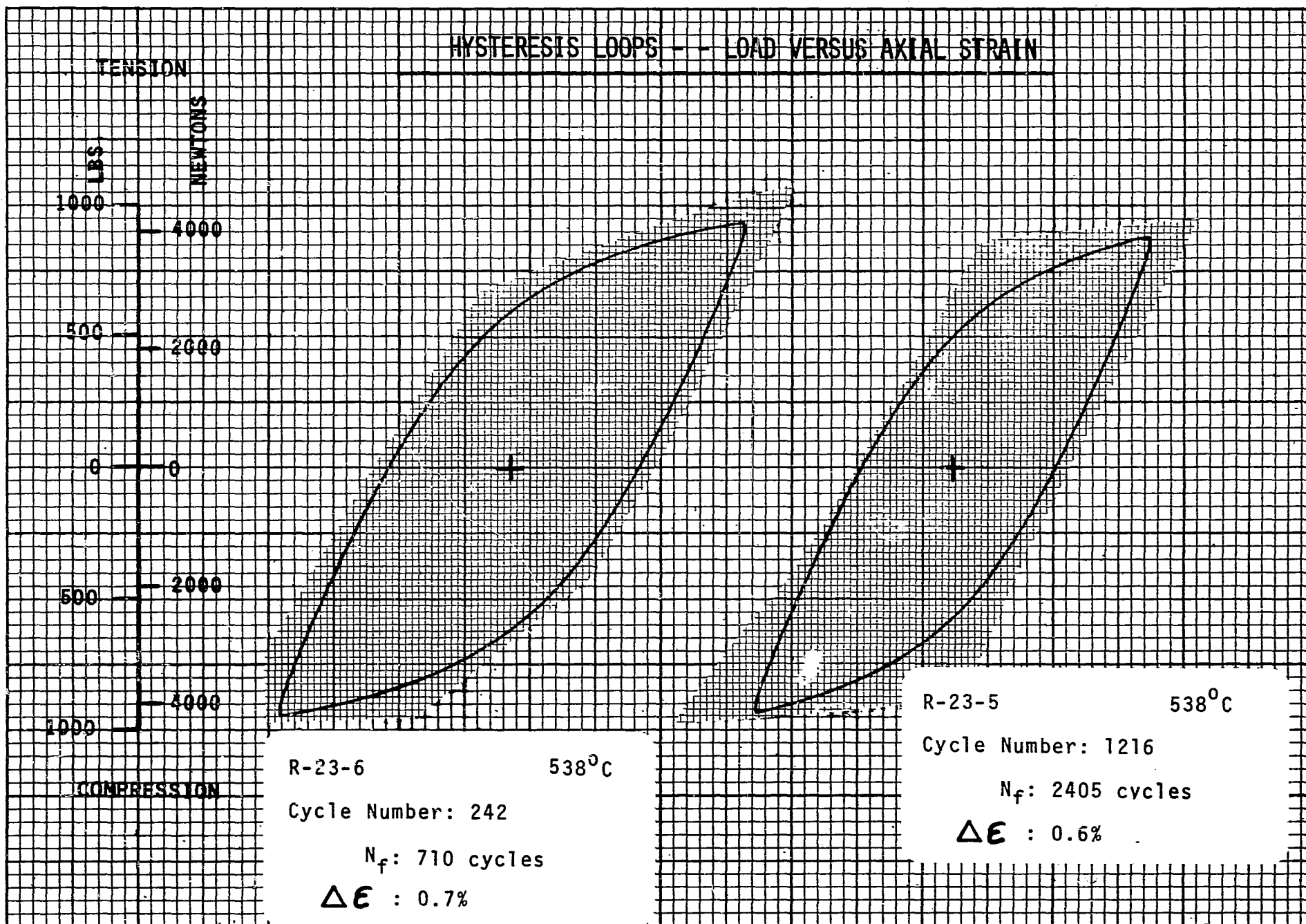


R-23-4 538°C
Cycle Number: 66
 N_f : 146 cycles
 $\Delta \epsilon$: 1.0%



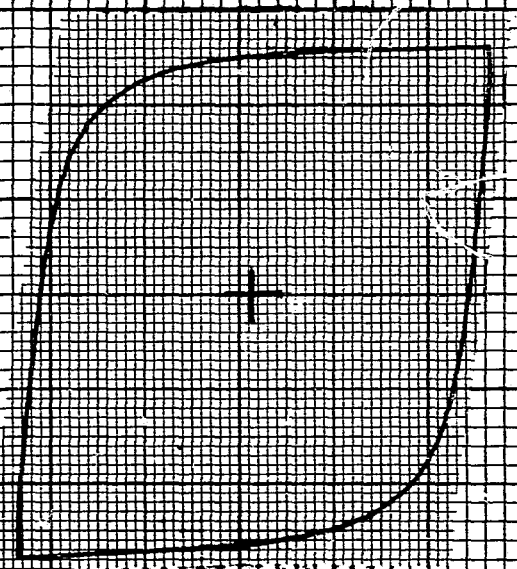
R-23-7 538°C
Cycle Number: 221
 N_f : 501 cycles
 $\Delta \epsilon$: 0.84%

Figure 274



HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION
LBS
NEWTONS
1000
500
0
500
1000
4000
2000
2000
4000
COMPRESSION

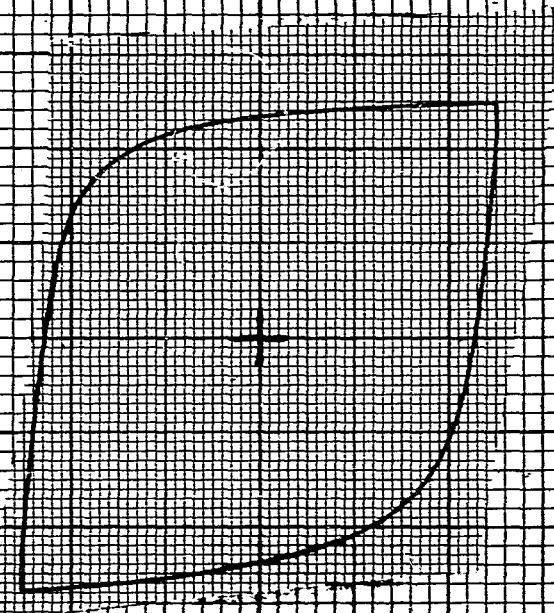


R-25-6 538°C

Cycle Number: 8

N_f : 58 cycles

$\Delta \epsilon$: 2.0%



R-25-7 538°C

Cycle Number: 57

N_f : 109 cycles

$\Delta \epsilon$: 2.0%

Figure 276

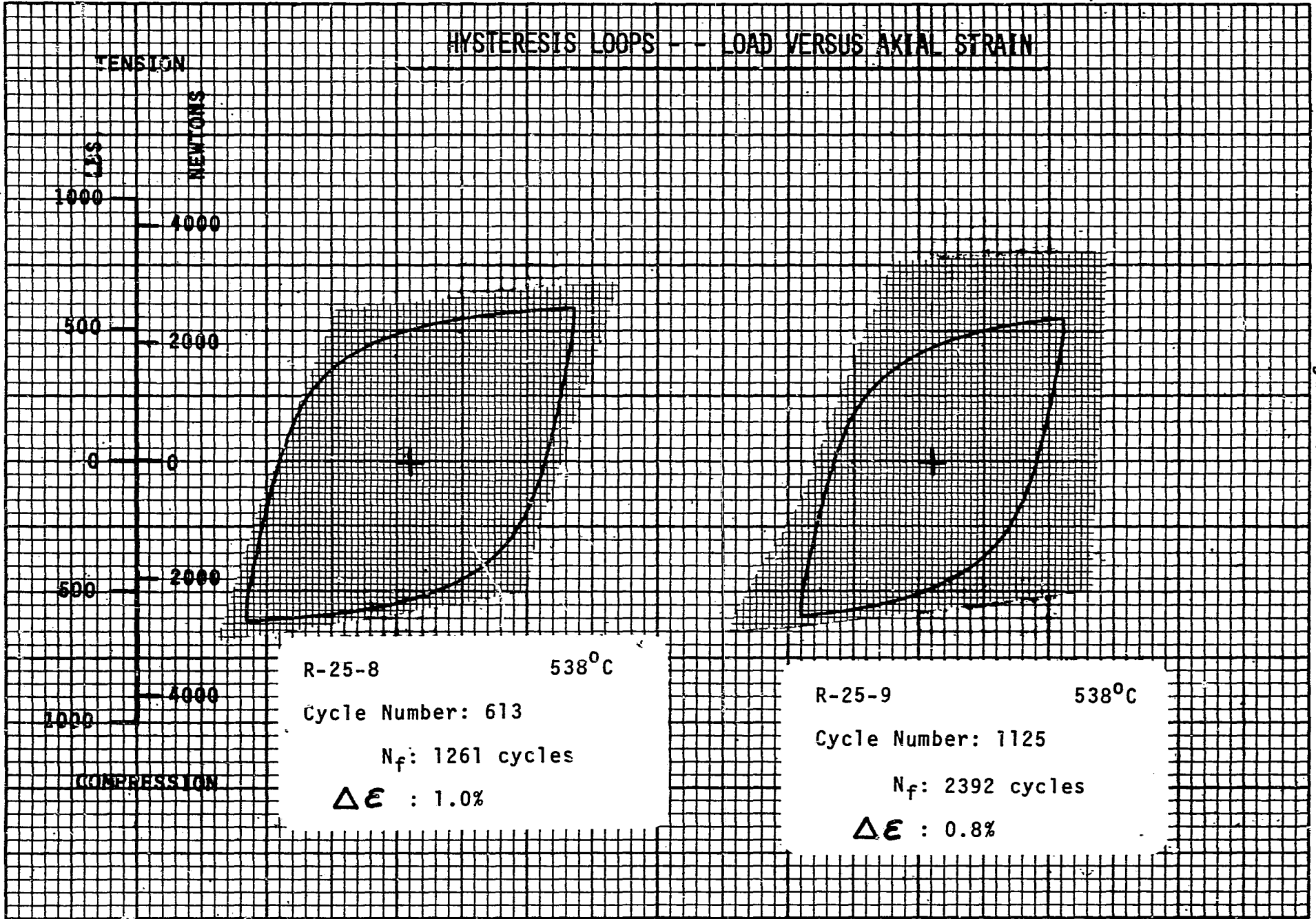
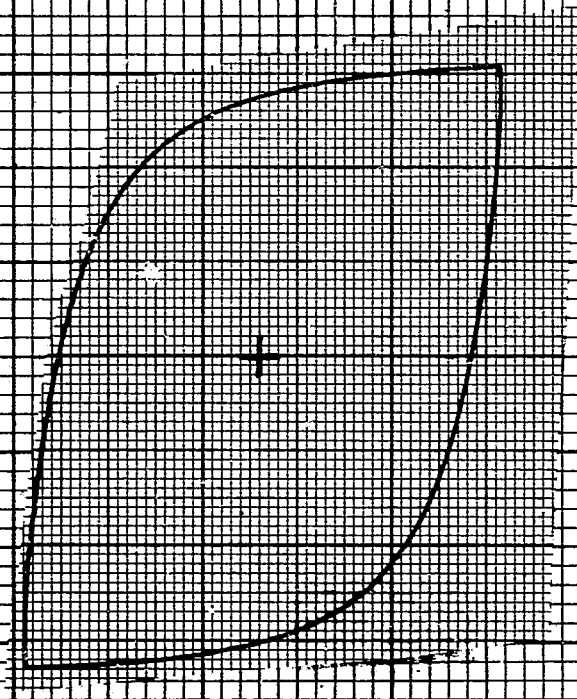


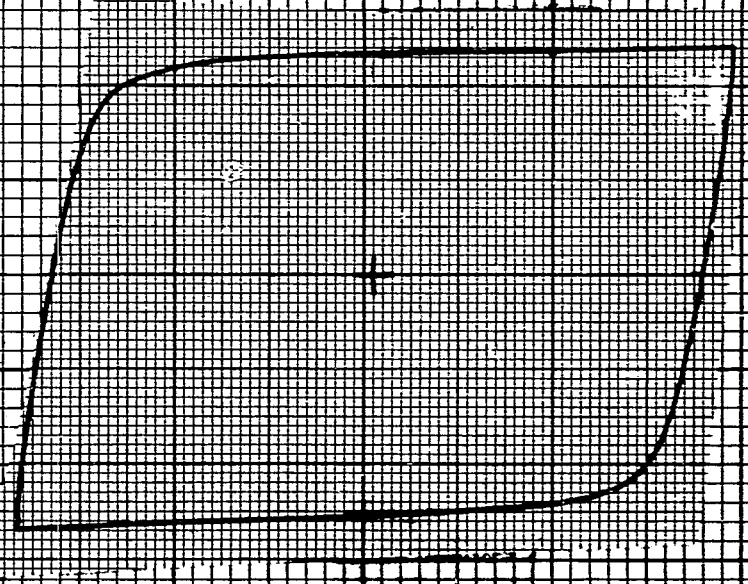
Figure 277

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

TENSION
LBS.
NEWTONS
1000
500
0
500
1000
4000
2000
0
2000
4000



R-26-2 538^oC
Cycle Number: 45
N_f: 109 cycles
 ΔE : 2.0%



R-22-18 538^oC
Cycle Number: 64
N_f: 120 cycles
 ΔE : 3.0%

COMPRESSION

Figure 278

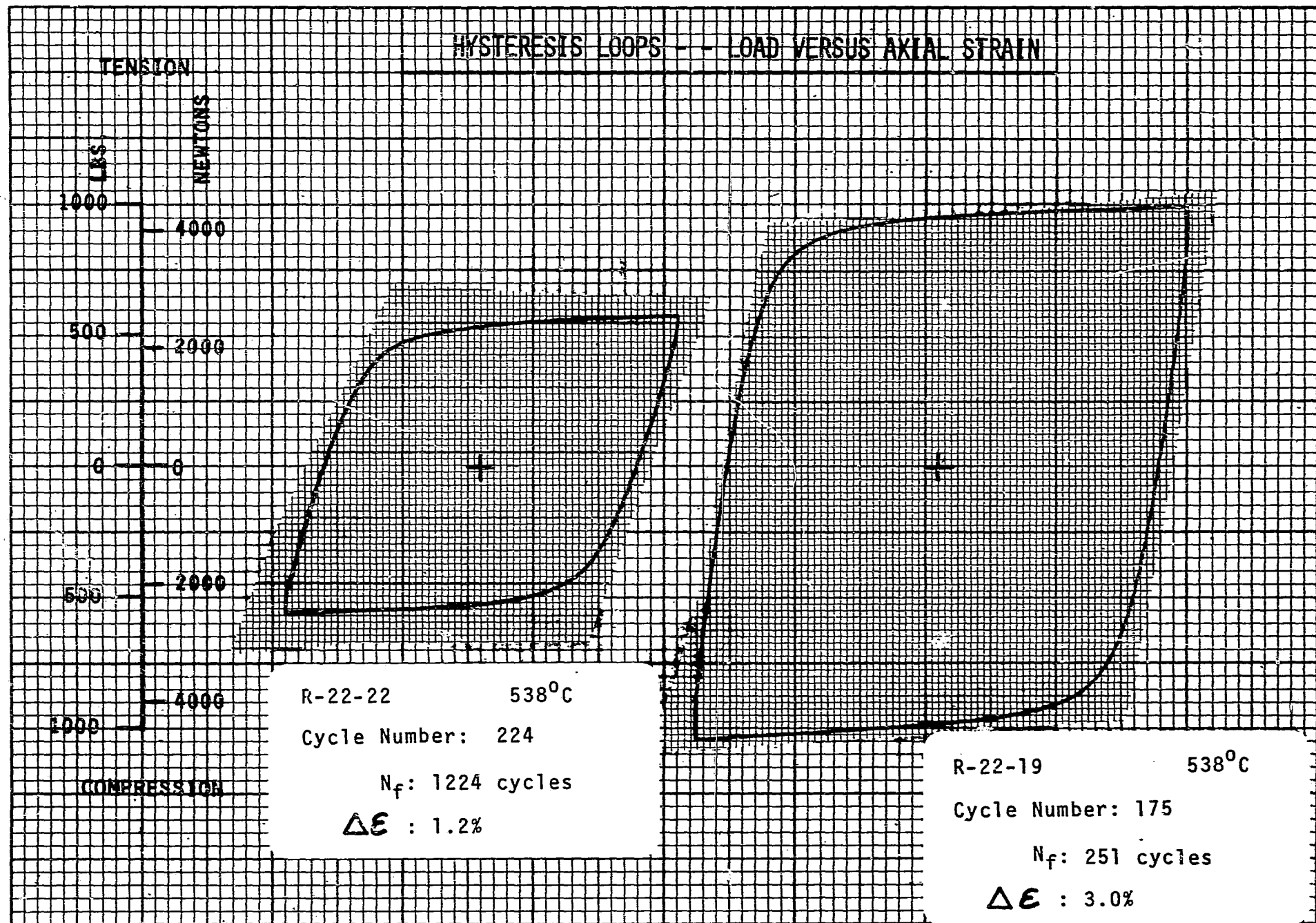


Figure 279

24

HYSTERESIS LOOPS -- LOAD VERSUS AXIAL STRAIN

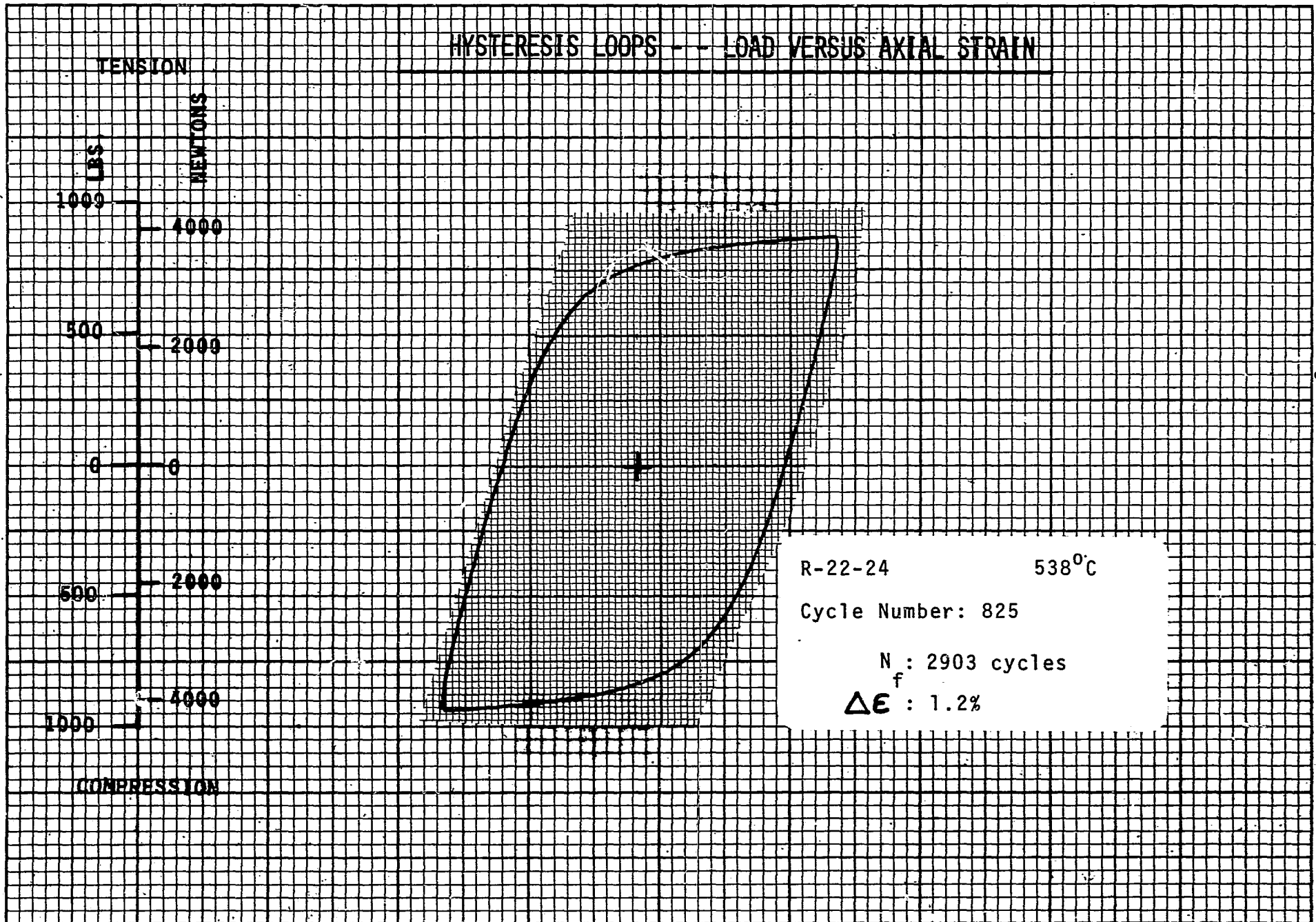


Figure 280

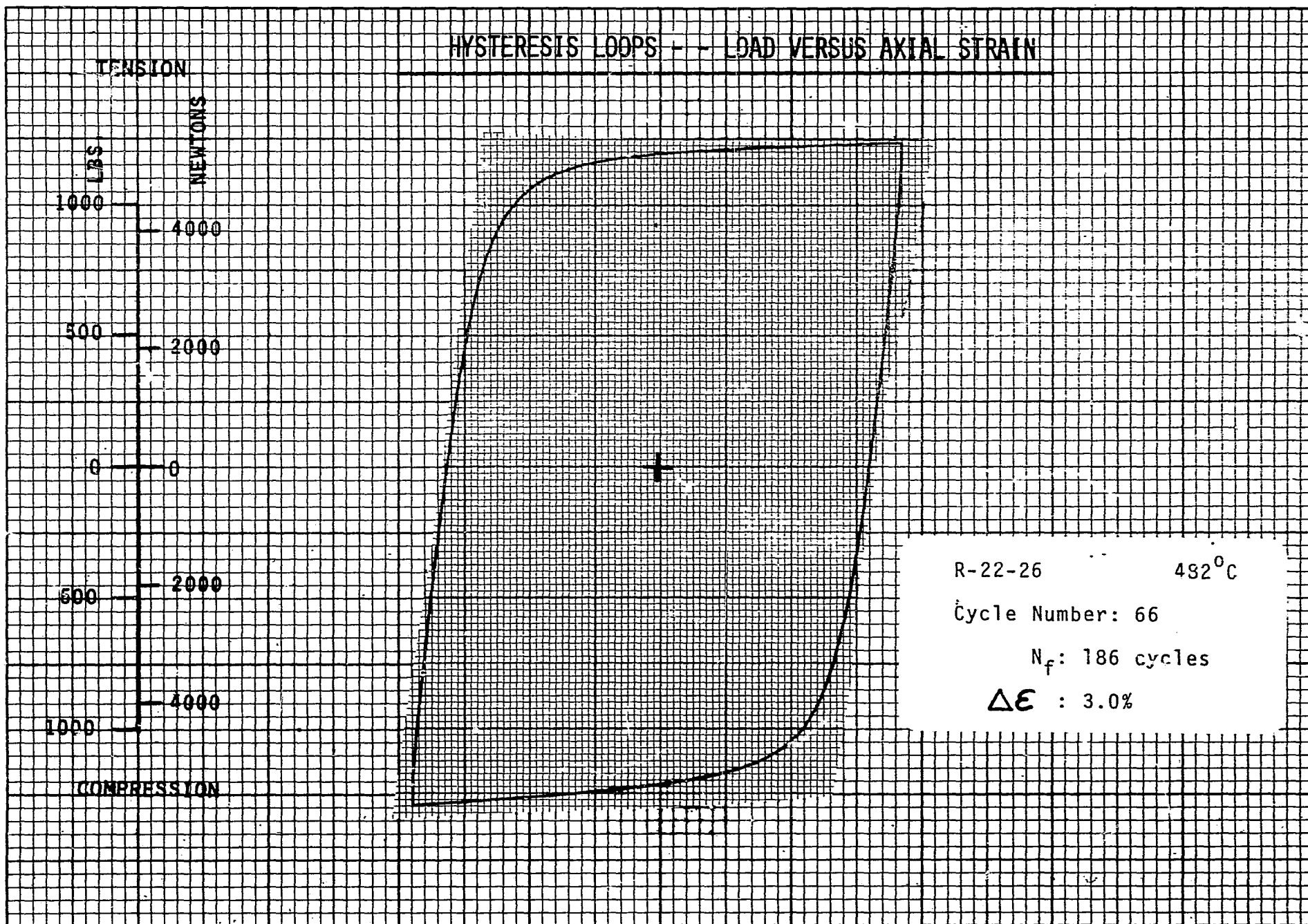


Figure 281

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

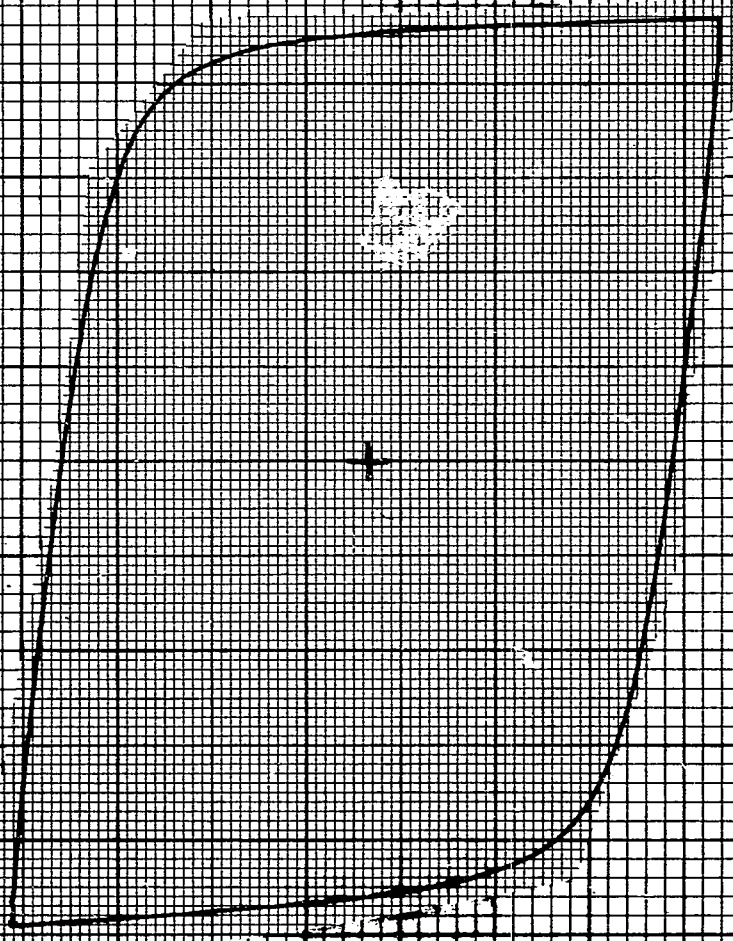
TENSION

1000
500
0
500
1000

NEWTONS

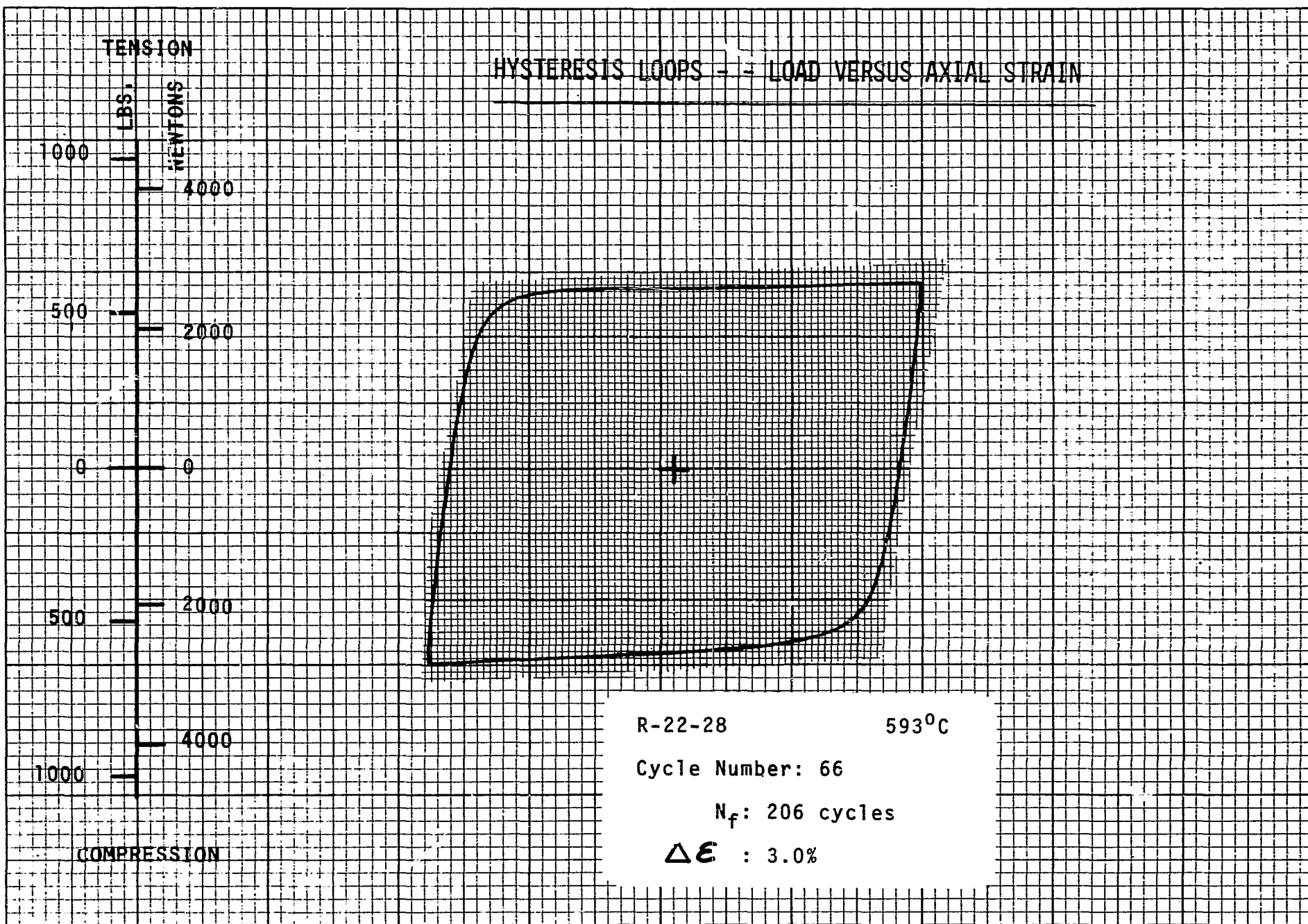
4000
2000
0
2000
4000

COMPRESSION



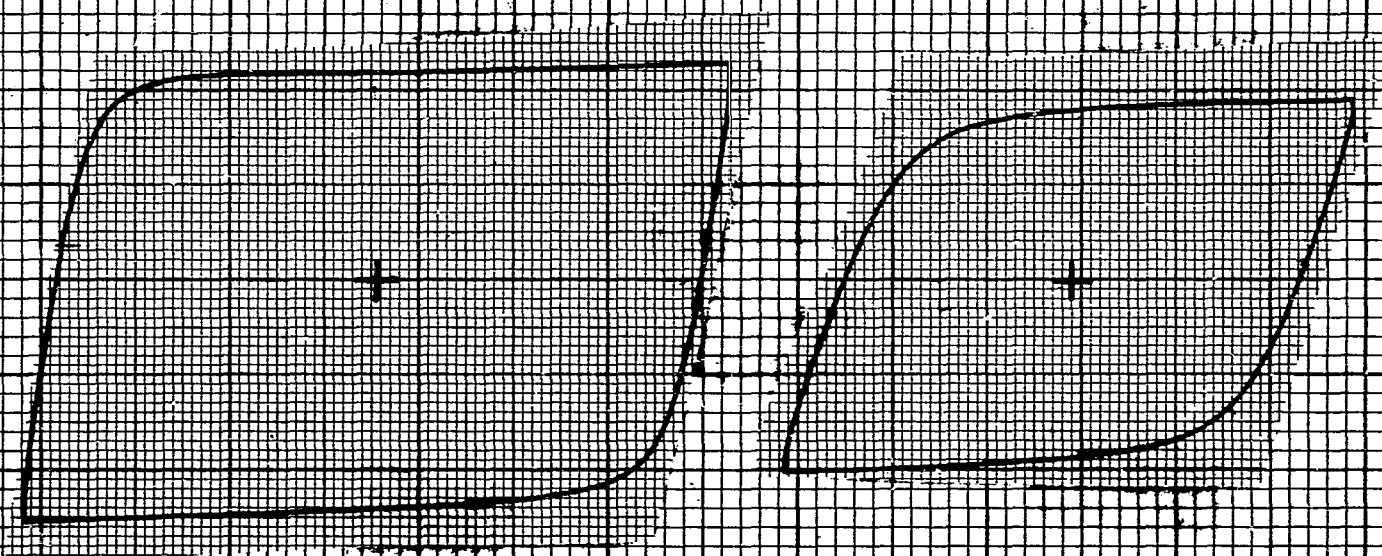
R-22-27 482°C
Cycle Number: 105
N_f: 205 cycles
 $\Delta \epsilon$: 3.0%

Figure 282



HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN

TENSION
 LBS
 NEWTONS
 1000
 500
 0
 500
 1000
 4000
 2000
 0
 2000
 4000

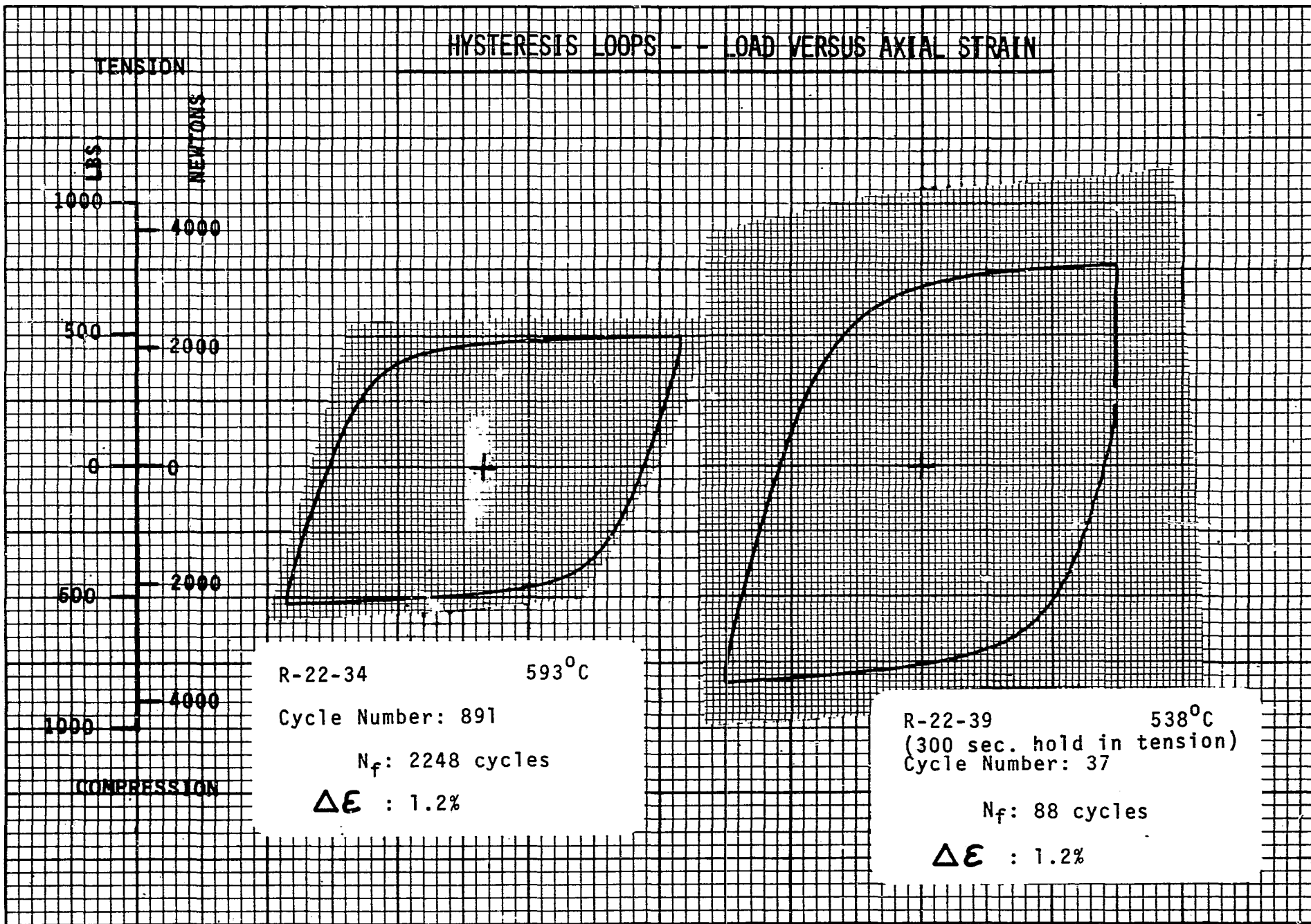


COMPRESSION

R-22-29 593°C
 Cycle Number: 78
 N_f: 192 cycles
 Δε : 3.0%

R-22-33 593°C
 Cycle Number: 1450
 N_f: 2893 cycles
 Δε : 1.2%

Figure 284



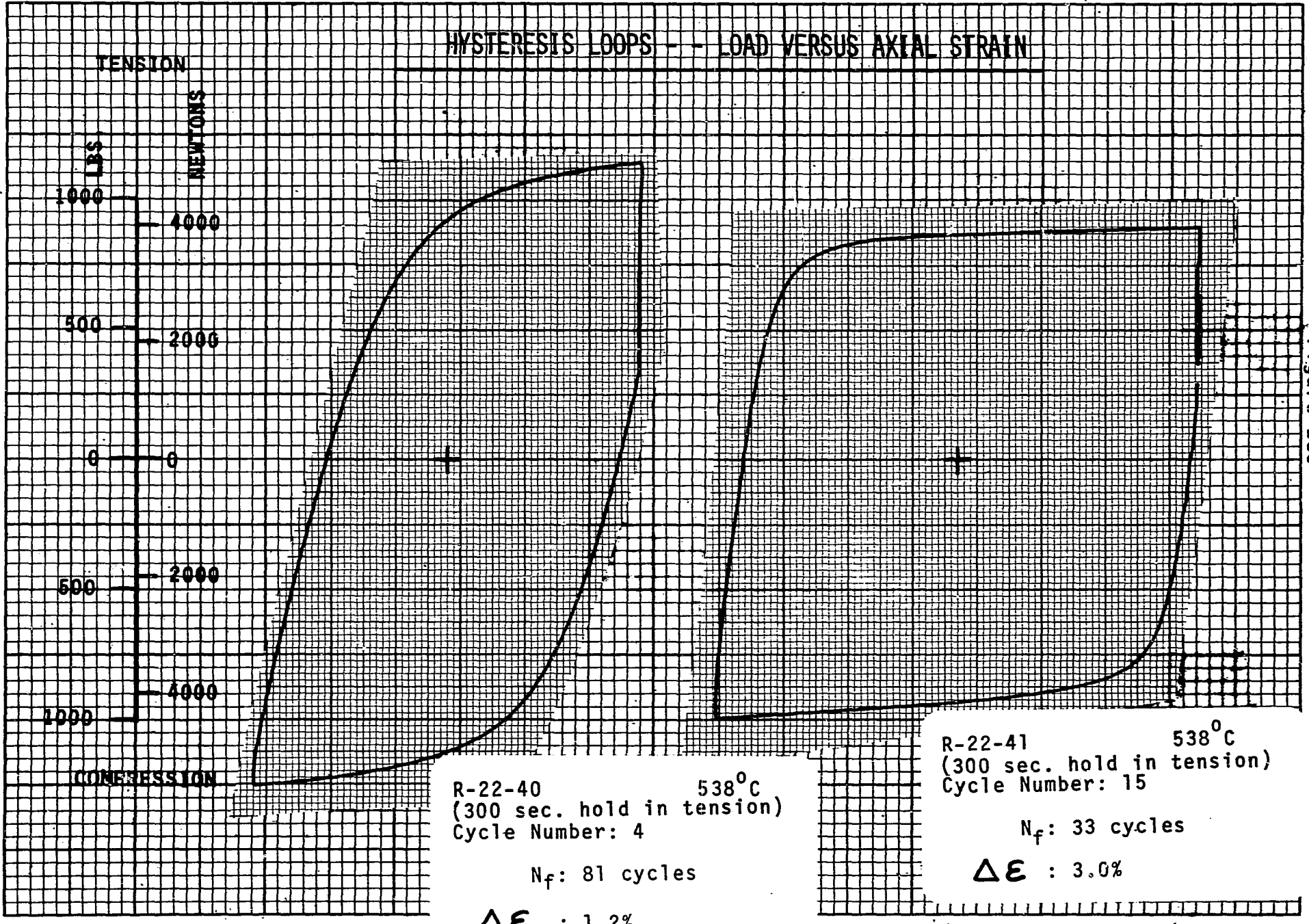


Figure 286

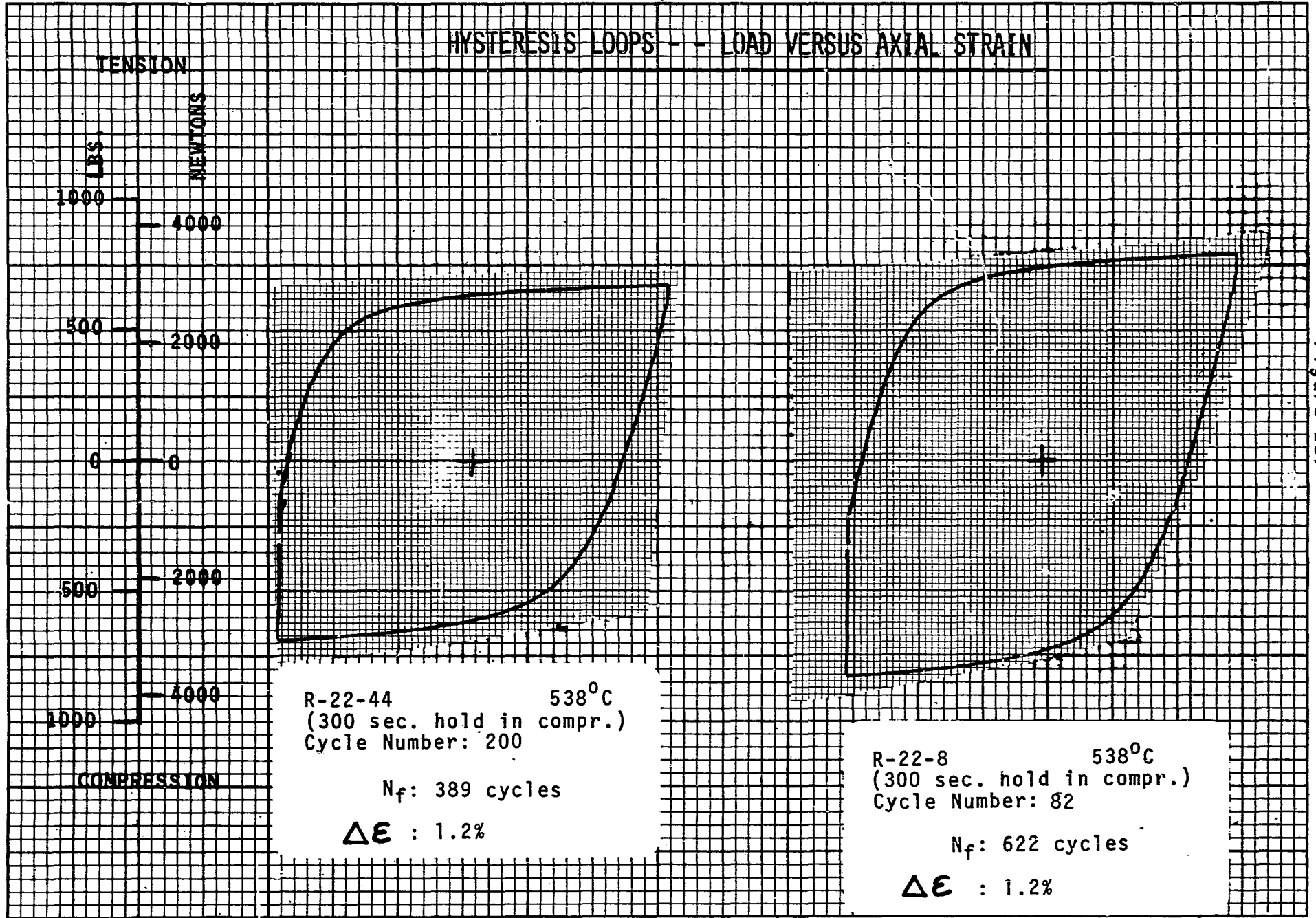
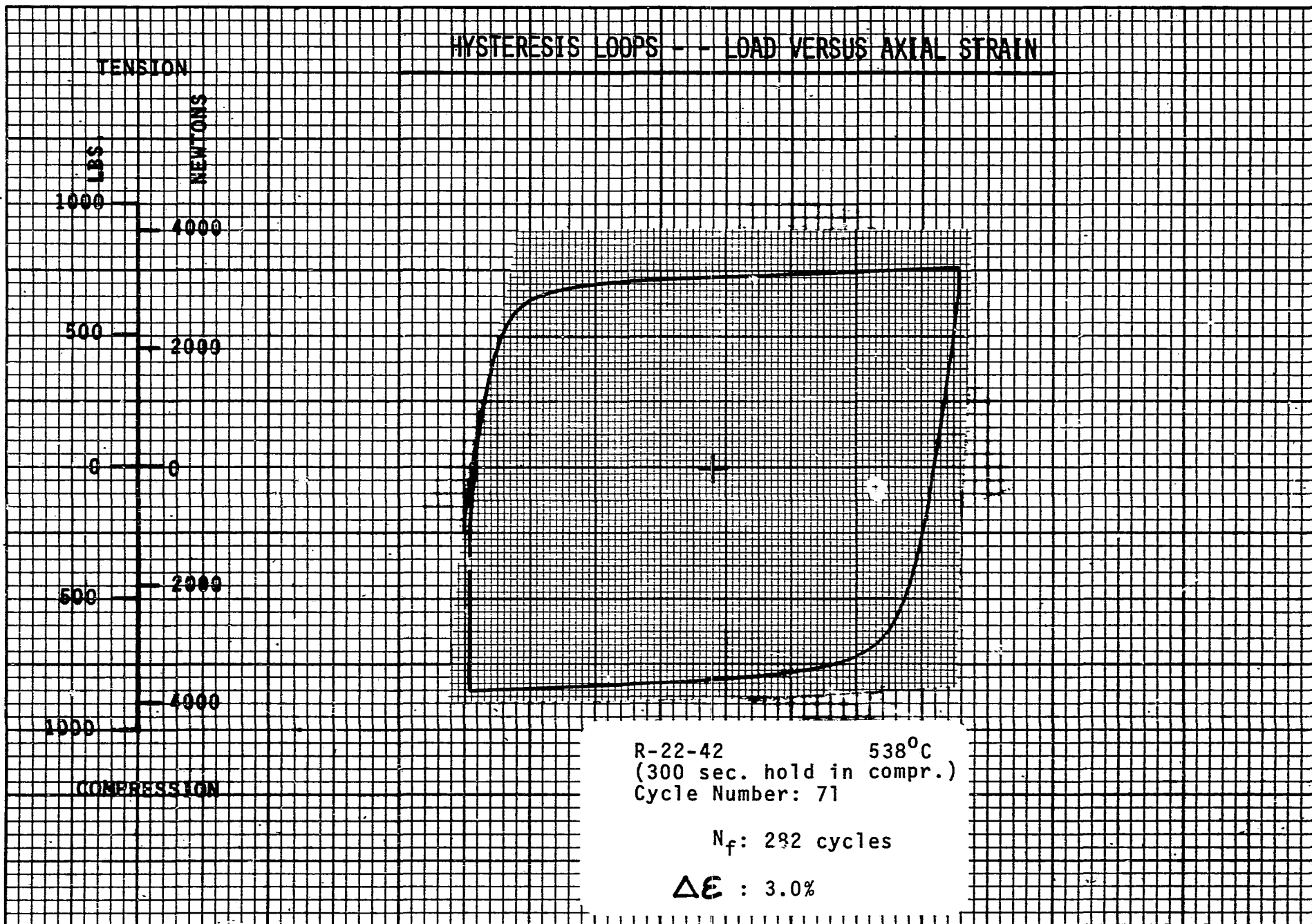


Figure 287

HYSTERESIS LOOPS - - LOAD VERSUS AXIAL STRAIN



R-22-42 538°C
(300 sec. hold in compr.)
Cycle Number: 71

N_f : 282 cycles

ΔE : 3.0%

Table 2 - Values of N_5 and N_f for all Tests

(N_5 is the number of cycles to a five percent load range drop and N_f is the number of cycles to complete separation of the specimen)

<u>Specimen Number</u>	<u>Temp., °C</u>	<u>Total Strain Range, %</u>	<u>Strain Rate, sec⁻¹</u>	<u>N_5, cycles</u>	<u>N_f, cycles</u>
<u>Zirconium-copper, annealed</u>					
R-0-14	538	2.0	2×10^{-3}	680	1512
R-0-15	↓	1.5	↓	1200	4188
R-0-25	↓	3.5	↓	220	283
R-0-17	↓	3.0	↓	190	307
R-0-19	↓	1.7	↓	1000	2300
R-0-21	↓	2.5	↓	410	418
<u>Zirconium-copper, ½ Hard</u>					
R-1-3	538	2.5	2×10^{-3}	320	524
R-1-4	↓	1.6	↓	800	1088
R-1-9	↓	3.5	↓	170	562
R-1-10	↓	3.5	↓	180	447
R-1-11	↓	1.2	↓	4900	5590
R-1-12	↓	1.35	↓	2100	3660
<u>Zirconium-copper, ½ Hard</u>					
R-2-2	538	3.0	2×10^{-3}	1120	1615
R-2-4	↓	5.0	↓	280	366
R-2-5	↓	4.0	↓	480	552
R-2-10	↓	2.8	↓	780	1055
R-2-11	↓	2.0	↓	720	1239
R-2-6	↓	2.0	↓	1400	2051
R-2-12	↓	1.7	↓	1770	1770
R-2-13	↓	1.5	↓	2400	2453
<u>Tellurium-copper, ½ Hard</u>					
R-3-10	538	1.6	2×10^{-3}	385	390
R-3-13	↓	2.0	↓	116	117
R-3-1	↓	1.2	↓	390	462
R-3-2	↓	0.8	↓	970	1179
R-3-4	↓	1.0	↓	660	802
R-3-5	↓	0.5	↓	3900	3908
<u>Chromium-copper, SA and aged</u>					
R-4-1	538	2.0	2×10^{-3}	110	147
R-4-2	↓	1.6	↓	290	354
R-4-3	↓	1.4	↓	530	605

<u>Specimen Number</u>	<u>Temp., °C</u>	<u>Total Strain Range, %</u>	<u>Strain Rate, sec⁻¹</u>	<u>N₅, cycles</u>	<u>N_f, cycles</u>
<u>Chromium-copper, SA and aged</u>					
R-4-4	538	1.0	2 x 10 ⁻³	1600	1823
R-4-5	↓	1.2	↓	970	1102
R-4-8		0.9		3500	3648
<u>OFHC Copper, Hard</u>					
R-5-2	538	1.6	2 x 10 ⁻³	280	292
R-5-11	↓	2.0	↓	160	195
R-5-1		1.0		-	679
R-5-4		0.8		-	1295
R-5-3		1.2		450	453
R-5-5		0.6		2800	3606
<u>OFHC Copper, ½ Hard</u>					
R-6-1	538	1.6	2 x 10 ⁻³	82	85
R-6-3	↓	0.6	↓	630	691
R-6-4		0.7		300	418
R-6-5		2.0		40	56
R-6-8		0.5		900	1358
R-6-9		1.0		180	200
R-6-11		1.0		200	303
<u>OFHC Copper, Annealed</u>					
R-7-3	538	2.0	2 x 10 ⁻³	75	126
R-7-4	↓	1.5	↓	230	269
R-7-1		1.2		400	437
R-7-2		1.0		660	710
R-7-7		0.8		1100	1313
R-7-9		0.7		1300	1613
<u>Silver, As-drawn</u>					
R-8-7	538	3.0	2 x 10 ⁻³	-	344
R-8-5	↓	2.5	↓	590	603
R-8-9		1.2		1700	1902
R-8-3		1.0		2500	2620
R-8-2		2.0		850	928
R-8-10		1.5		1350	1381
<u>Zr-Cr-Mg Copper, SA, CW and Aged</u>					
R-9-11	538	2.0	2 x 10 ⁻³	700	843
R-9-4	↓	3.0	↓	270	357
R-9-1		2.25		360	500
R-9-3		2.5		300	346

Table 2 continued

<u>Specimen Number</u>	<u>Temp., °C</u>	<u>Total Strain Range, %</u>	<u>Strain Rate, sec⁻¹</u>	<u>N₅, cycles</u>	<u>N_f, cycles</u>
<u>Zr-Cr-Mg Copper, SA, CW and aged</u>					
R-9-8	538	1.4	2 x 10 ⁻³	1500	2000
R-9-2	↓	1.2	↓	1250	1317
R-9-9		0.9		4500	6670
<u>Electroformed Copper, 30-35 ksi</u>					
R-10-3	538	2.0	2 x 10 ⁻³	30	148
R-10-4	↓	1.6	↓	25	38
R-10-1		0.8		450	1542
R-10-8		1.2		60	72
R-10-5	↓	1.0	↓	250	512
R-10-7		0.75		1000	1866
<u>Co-Be-Zr Copper, SA and aged</u>					
R-13-5	538	2.0	2 x 10 ⁻³	70	90
R-13-13	↓	1.2	↓	350	644
R-13-7		1.5		130	212
R-13-10		1.0		450	680
R-13-6	↓	0.8	↓	700	1615
R-13-8		0.7		1800	3623
<u>Zirconium-Copper, ½ Hard</u>					
R-2-26	482	6.1	2 x 10 ⁻³	100	204
R-2-27	↓	1.41	↓	2000	2985
R-2-28		6.1		98	176
R-2-29	↓	1.41	↓	1900	2135
R-2-30		5.0		145	265
R-2-31	593	1.4	2 x 10 ⁻³	3100	3380
R-2-32	↓	5.0	↓	230	346
R-2-33	↓	1.5	↓	1750	2008
R-2-34		5.0		175	234
R-2-36	538	5.0	4 x 10 ⁻⁴	170	234
R-2-37	↓	1.4	↓	-	1613
R-2-39		5.0		200	238
R-2-40		1.4		3000	3693
R-2-35		5.0	1 x 10 ⁻²	-	245
R-2-38		1.4	↓	2500	5431
R-2-41		1.4	↓	2600	5215
R-2-42		5.0	↓	230	462
R-2-43		5.0	200 sec, T	170	211
R-2-44		5.0	200 sec, T	165	190
R-2-45		5.0	200 sec, C	253	253
R-2-46	↓	5.0	200 sec, C	210	262

Table 2 continued

<u>Specimen Number</u>	<u>Temp., °C</u>	<u>Total Strain Range, %</u>	<u>Strain Rate, sec⁻¹</u>	<u>N₅, cycles</u>	<u>N_f, cycles</u>
<u>Zirconium-Copper, ½ Hard</u>					
R-2-47	538	1.4	56 sec,T	1100	1152
R-2-48	↓	1.4	56 sec,T	-	1062
R-2-49	↓	1.4	56 sec,C	-	1947
R-2-50	↓	1.4	56 sec,C	2800	3180
R-2-17	RT	3.0	5 x 10 ⁻³	-	696
R-2-51	↓	2.5	↓	1000	1102
R-2-52	↓	1.6	↓	-	2780
R-2-53	↓	4.0	↓	400	425
R-2-54	↓	1.4	↓	2900	3283
R-2-55	↓	3.0	↓	70	141
R-2-56	↓	1.4	↓	250	503
R-2-57	↓	0.9	3 x 10 ⁻³	750	1784
R-2-58	↓	2.0	5 x 10 ⁻³	130	279
R-2-59	↓	0.7	3 x 10 ⁻³	2000	5607
R-20-1	538	5.0	2 x 10 ⁻³	250	390
R-20-2	↓	1.4	↓	4000	5658
R-20-3	↓	2.0	↓	1200	1326
R-20-4	↓	3.0	↓	-	879
R-20-5	↓	1.6	↓	3400	3589
R-20-6	↓	1.8	↓	2600	2700
R-2-67	538	5.0	300 sec,T	-	218
R-2-68	↓	1.4	300 sec,T	-	1156
R-2-69	↓	4.62	300 sec,C	-	948
R-2-72	↓	1.4	300 sec,C	-	1224
R-2-73	↓	5.0	300 sec,T,C	-	234
R-2-61	538/260	3.5	8.75 x 10 ⁻⁵	240	284
R-2-62	538/260	3.5	8.75 x 10 ⁻⁵	-	188
<u>Narloy Z, Cent. cast, hot-rolled, solution annealed and aged</u>					
R-24-13	538	1.0	2 x 10 ⁻³	1150	1169
R-24-14	↓	2.0	↓	260	331
R-24-15	↓	1.2	↓	870	1126
R-24-16	↓	0.7	↓	3000	3601
R-24-17	↓	3.5	↓	92	99
R-24-18	↓	2.5	↓	200	253
R-24-19	↓	0.85	↓	1700	2469
R-24-22	↓	0.9	1 x 10 ⁻²	2800	3909
R-24-21	↓	0.9	↓	3230	3586
R-24-23	↓	2.6	↓	290	339
R-24-26	↓	2.6	↓	340	364
R-24-20	↓	0.9	4 x 10 ⁻⁴	860	1138
R-24-25	↓	0.9	4 x 10 ⁻⁴	880	1196

Table 2 continued

<u>Specimen Number</u>	<u>Temp., °C</u>	<u>Total Strain Range, %</u>	<u>Strain Rate, sec⁻¹</u>	<u>N₅, cycles</u>	<u>N_f, cycles</u>
<u>Narloy Z, Cent. cast, hot-rolled, solution annealed and aged</u>					
R-24-24	538	2.6	4×10^{-4}	120	154
R-24-27	538	2.6	4×10^{-4}	122	133
R-24-29	538	2.6	5.2×10^{-2}	474	474
R-24-30	538	2.6	5.2×10^{-2}	570	588
<u>Zirconium-Copper, ½ Hard</u>					
R-2-74	538	2.6	5.2×10^{-2}	2500	3132
R-2-75	538	2.6	5.2×10^{-2}	-	3480
<u>Narloy Z, Cent. cast, hot-rolled, solution annealed and aged</u>					
R-24-33	593	0.9	2×10^{-3}	1180	1253
R-24-34	593	2.6	↓	160	191
R-24-31	482	0.9	↓	2800	2950
R-24-32	482	2.6	↓	180	243
R-24-39	538	2.6	290 sec,T	63	102
R-24-40	↓	2.6	300 sec,T	55	75
R-24-41	↓	2.6	300 sec,C	220	353
R-24-42	↓	2.6	300 sec,C	220	337
R-24-38	↓	0.9	300 sec,T	190	262
R-24-37	↓	0.9	300 sec,T	195	317
R-24-43	↓	0.9	300 sec,C	1500	2981
R-24-45	↓	0.9	300 sec,C	1500	3392
<u>NASA 1-1A Copper Alloy, Aged</u>					
R-21-3	538	1.5	2×10^{-3}	1090	1101
R-21-4	↓	3.0	↓	265	270
R-21-5	↓	1.0	↓	-	4120
R-21-6	↓	5.0	↓	85	92
<u>NASA 1-1B Copper Alloy, As-received</u>					
R-22-4	538	2.0	2×10^{-3}	-	488
R-22-5	↓	4.0	↓	90	106
R-22-6	↓	1.5	↓	-	1122
R-22-7	↓	1.2	↓	1950	1992
<u>Glidcop AL-10</u>					
R-23-4	538	1.0	2×10^{-3}	110	146
R-23-7	↓	0.84	↓	420	501
R-23-6	↓	0.7	↓	620	710
R-23-5	↓	0.6	↓	2000	2405

<u>Specimen Number</u>	<u>Temp., °C</u>	<u>Total Strain Range, %</u>	<u>Strain Rate, sec⁻¹</u>	<u>N₅, cycles</u>	<u>N_f, cycles</u>
<u>Sputtered Zirconium-Copper, Annealed</u>					
R-25-6	538	2.0	2×10^{-3}	50	58
R-25-7	↓	2.0	↓	80	109
R-25-8	↓	1.0	↓	900	1261
R-25-9	↓	0.8	↓	1700	2392
<u>Sputtered Zirconium-Copper, As-sputtered</u>					
R-26-2	538	2.0	2×10^{-3}	100	109
<u>NASA 1-1B Copper Alloy, As-received</u>					
R-22-18	538	3.0	4×10^{-4}	-	120
R-22-22	↓	1.2	4×10^{-4}	-	1224
R-22-19	↓	3.0	1×10^{-2}	190	251
R-22-24	↓	1.2	1×10^{-2}	2450	2903
R-22-26	482	3.0	2×10^{-3}	150	186
R-22-27	482	3.0	↓	155	205
R-22-28	593	3.0	↓	180	206
R-22-29	↓	3.0	↓	182	192
R-22-33	↓	1.2	↓	2880	2893
R-22-34	↓	1.2	↓	1900	2248
R-22-39	538	1.2	300 sec,T	77	88
R-22-40	↓	1.2	300 sec,T	67	81
R-22-41	↓	3.0	300 sec,T	29	33
R-22-44	↓	1.2	300 sec,C	389	389
R-22-8	↓	1.2	300 sec,C	-	622
R-22-42	↓	3.0	300 sec,C	240	282