

THE NATA CODE - PROGRAMMER'S MANUAL

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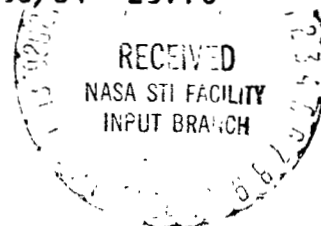
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16. Abstract The NATA code is a computer program for calculating quasi-one-dimensional gas flow in axisymmetric and two dimensional nozzles and rectangular channels. The code also computes stagnation point conditions on axisymmetric or two-dimensional models and the conditions on blunt wedges inserted into the flow. The theory and analysis underlying the calculations have been documented in Volume I of this final report. Code inputs, outputs, and precoded data have been defined in Volume II, which is a user's manual. The present, third volume of the final report is a programmer's manual for the code. It provides a listing of the Fortran IV source program; a complete glossary of Fortran symbols; a discussion of the purpose and method of operation of each subroutine (including mathematical analyses of special algorithms); and a discussion of the operation of the code on IBM/360 and UNIVAC 1108 systems, including required control cards and the overlay structure used to accommodate the code to the limited core size of the 1108. In addition, this volume provides similar information to document the programming of the NOZFIT code, which is employed to set up nozzle profile curvefits for use in NATA.					
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PREFACE

This is the third volume of a final report documenting the development of a computer program (NATA) for calculating the flow in arc-heated wind tunnels and the conditions on models tested in such reentry simulation facilities. Volume I of the report provides a general summary of the code's capabilities and presents the theory and analysis underlying its operation. Volume II is a user's manual which defines NATA's inputs, outputs, and precoded data on standard gas models and nozzle geometries. The present volume is a programmer's manual for NATA. It documents the programming of NATA in sufficient detail to support code modifications, and provides information needed for running the code on the UNIVAC 1108 under the EXEC II system, or on the IBM 360/75 under HASP II.

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THE NATA CODE - PROGRAMMER'S MANUAL

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1. INTRODUCTION

The NATA code is a computer program for solving problems of steady, quasi-one-dimensional gas flow in nozzles. The code's capabilities, and the theory and analysis underlying its operation, have been documented in the first volume of this report (ref. 1). The inputs, outputs, and precoded data on standard gas models and nozzle geometries have been defined in the second volume (ref. 2). The present, third volume provides information required for running NATA on UNIVAC 1108 and IBM 360 computers, and documents the programming of NATA so that programmers and computer-oriented users will be able to evaluate the techniques used, maintain the code in the face of operating-system changes, convert it for use on other types of computer, introduce modifications and correct errors.

Section 2 provides a general description of NATA, including a catalog of its subroutines with a statement of function and a list of calling routines for each. This section also gives sample control-card decks for running the code on the UNIVAC 1108 under the EXEC II system, and on the IBM 360/75 under HASP II and CS 21.

Section 3 discusses each subroutine, explaining its purpose in the code and its method of operation. Many special algorithms are analyzed. Where appropriate, flowcharts are used to display the logic and clarify the methods used. The Fortran variables stored in common blocks and used in the individual routines are all defined in a glossary of symbols (Section 4). Section 5 lists the subroutines in which each common variable is set, used, or referenced as a subroutine argument. Finally, a complete listing of the source program is given in Section 6.

An appendix provides similar documentation of the NOZFIT code, a much smaller program provided for setting up nozzle profile curvefits for use in NATA. A user's manual for NOZFIT is included in Volume II (ref. 2). The appendix includes documentation of a general-purpose plotting subroutine (GRAPH) for producing plots on S-C 4020, S-C 4060, or other plotting equipment using the NAA software package for the SC-4020 (ref. 3).

2. STRUCTURE AND OPERATION

2.1 Brief Description of the NATA Code

NATA is a Fortran IV program consisting of a main program, two block data routines, and 66 subroutines. The source deck contains approximately 8500 cards. The program exists in two versions, one for use on the IBM 360 system, the other for the UNIVAC 1108. The IBM version is entirely in double precision, whereas the UNIVAC version is a single precision program with some double precision arrays and subroutines. One version can be converted into the other by inserting or removing most of the IMPLICIT REAL*8 (A-H, ϕ -Z) cards which type all floating-point variables as double precision in the IBM 360 version.*

The IBM 360 version requires about 410K bytes of core storage, including buffers. The UNIVAC version is run on the 1108 using overlay, and requires about 123000₈ or 42500₁₀ words of storage. It fits into the two-bank processors at NASA/JSC with about 2000 words to spare.

2.2 Subroutines and Linkage

A list of the routines of NATA is presented below. Names of entry points are included, and are enclosed in parentheses to distinguish them from subroutine and function names. The function performed by each routine or entry point is described briefly, and in each case a complete list of calling routines is included. The order of the list is as follows: main program; block data routines; subroutines, functions, and entry points in alphanumeric order.

*Also, the IBM version contains two extra subroutines, EXP and RESET, for reasons explained below.

MAIN Main program; calls other routines to perform the required calculations.

BLKD1 BLOCK DATA routine containing precoded data on elements, species, reactions, gas models, and nozzle geometries.

BLKD2 BLOCK DATA routine containing precoded data on transport-property cross sections.

AESØLN Calculates the effective area ratio A_e and its derivative dA_e/dx from data on nozzle geometry and the boundary layer displacement thickness.* Called by GEØM.

AGSØLN Calculates the geometric area ratio A_g and the axial coordinate x from data on the effective area ratio A_e and the boundary layer displacement thickness. Called by FRØZEN, EQUIL, and AXFIT.

AXFIT Calculates the axial coordinate x corresponding to a given value of the effective area ratio A_e in the nonequilibrium solution by the perturbation method. Called by NØNEQ.

AXSECT For a given list of species pairs, adds those pairs whose cross sections have not previously been specified to a particular step of the transport cross section calculation. Called by XSECT.

BLAYER Calculates boundary layer momentum and displacement thicknesses, heat flux, and shear stress. Called by FRØZEN, EQUIL, and BICALL.

BICALL Sets up and executes calls to BLAYER in the nonequilibrium solution. Called by DERIVS and PRTA.

*Throughout this report, the mathematical notation is as defined in Volume I (ref. 1).

BXSECT	Searches a parameter array for references to the indices of species pairs used in the cross section calculation; also determines the correspondence between entries in the parameter array and the steps in the cross section calculation. Called by XSECT.
CJMM	Computes the species production terms $P_i \chi_i$ in the nonequilibrium solution; in runs with an electronic nonequilibrium gas model, also computes the radiative loss and the energy transfer to the electron gas. Called by DERIVS.
CXSECT	Determines the correspondence between species pairs in the master list of species and those in the species list for the current case. Called by XSECT.
DERIVS	Organizes the calculations of rates of change of the species concentrations and other flow variables in the nonequilibrium solution. Called by NØNEQ and RNKT.
DSMSØL	Performs simultaneous solution of a system of linear, inhomogeneous equations by calling subroutine SIMQ. Called by EQCALC, EXACT, NEWRAP, NØNEQ, and PERT.
DUMP	Prints name of routine in which an error has been detected and sets an error indicator to terminate the case. Called by AGSØLN, DSMSØL, EQCALC, FINDX, GEØM, GEØMAR, MATINV, NEWRAP, NØNEQ, PRØP, RESTMP, and WEDGE.
DUMPEA	Prints a large diagnostic dump of common data when a case is terminated by a call to DUMP because of an error. Called by MAIN.
ELCØND	Computes the electrical conductivity of the gas mixture. Called by TRANSP.

ELTIME	Prints elapsed time since the beginning of the run and time since the last previous time message. Called by MAIN and MØDEL.
EPART	Computes parameters for radiative energy loss and energy transfer to the electrons for a specified reaction. Called by CØMM.
EQCALC	Computes mole fractions and thermodynamic conditions for thermochemical equilibrium at specified temperature and pressure. Called by INTA and MØDEL.
(EQUIL)	Entry point in subroutine FRØZEQ; organizes the calculation of the equilibrium flow solution. Called by MAIN.
EXACT	Computes the derivatives of the species concentrations and other flow variables in the non-equilibrium solution. Called by DERIVS.
EXP	In the IBM 360 version, the double precision exponential function e^x with underflow suppression; included to avoid underflow messages. Not used in UNIVAC version. Called by BLAYER, CØMM, EQCALC, NEWRAP, ØUT1, PRØP, RNKT, THERM, and WEDGE.
FINDX	Solves for the axial coordinate x corresponding to a given geometric area ratio A_g in the upstream or downstream region. Called by AGSØLN, AXFIT, FRØZEQ, MAIN, and NEXTMP.
(FINDXC)	Entry point in subroutine FINDX; solves for the axial coordinate x corresponding to a given half width of one of the profiles in a channel. Called by NEXTMP.
(FRØZEN)	Entry point in subroutine FRØZEQ; organizes the calculation of the frozen flow solution. Called by MAIN.

FRØZ EQ Organizes the frozen and equilibrium flow solutions. The subroutine name is not referenced in NATA; called only through the entry points EQUIL and FRØZEN.

GEØM Organizes the geometry calculations for the non-equilibrium solution. In the direct integration, it computes A_e and $d \ln A_e / dx$; in the inverse method, it computes ρ and $d \ln \rho / dx$. Called by CØMM.

GEØMAR Computes the geometric area ratio A_g and its derivative $d A_g / dx$ at a given axial coordinate x . Called by AGSØLN, BLAYER, FINDX, GEØM, ØUT1, and THRØAT.

(GMAR) Entry point of subroutine GEØMAR; computes the profile ordinate y at a specified axial coordinate x . Called by CØMM, ØUT1, and RADIUS.

(GMAR2) Entry point of subroutine GEØMAR; computes the profile ordinates y, z at a specified axial coordinate x in a channel. Called by AGSØLN, CØMM, ØUT1, and RADIUS.

(GMAR3) Entry point of subroutine GEØMAR; computes the profile ordinates, y, z and their derivatives $dy/dx, dz/dx$ at a specified axial coordinate x in a channel. Called by AESØLN, BLAYER, and FINDX.

INGAS Sets up arrays used in the chemical description of the gas mixture. Called by MAIN.

INIT Initializes same control parameters and nondimensionalizes the species thermochemical data using the reservoir temperature. Called by MAIN and RESTMP.

(INTA) Entry point of subroutine INGAS; organizes the calculation of equilibrium conditions in the upstream reservoir for given reservoir temperature and pressure. Called by MAIN and RESTMP.

KANDMU Calculates quantities proportional to the viscosity and translational thermal conductivity of the gas mixture. Called by TRANSP.

KINT Computes quantities used in the transport property calculations. Called by TRANSP.

LIST Prints the "problem summary" portion of the output for each case. Called by MAIN.

MATINV Computes the inverse A_{ki} of the matrix $\bar{\alpha}_{ij}$ specifying the elemental composition of the independent species. Called by INGAS.

MØDEL Computes stagnation conditions on models inserted into the flow. Called by FRØZEQ, NØNEQ, and PRTA.

NEWRAP Computes the equilibrium flow conditions at specified temperature and entropy. Called by FRØZEQ, MAIN, NØNEQ, and NRMAX.

NEXTMP Determines the axial coordinate of the next location at which model condition calculations are to be done. Called by FRØZEQ and NØNEQ.

NØNEQ Organizes the nonequilibrium flow solution. Called by MAIN.

NRMAX Calculates the temperature and other conditions at the sonic point in an equilibrium flow starting at specified upstream reservoir conditions. Called by MAIN and RESTMP.

ØUT Prints a list of definitions of output variables. Called by MAIN.

ØUT1	Prints conditions in the upstream reservoir and at the sonic points of the frozen and equilibrium solutions. Called by MAIN.
(ØUT2)	Entry point of subroutine ØUT1; prints the conditions at each point in the flow solutions. Called by FRØZEQ and PRTA.
PERT	Computes the perturbations in the flow variables during the nonequilibrium solution by the perturbation method. Called by NØNEQ.
PIØMEG	Computes the Maxwell-averaged electron-neutral momentum transfer cross section for given electron temperature T_e . Called by CØMM.
PRØP	Computes the conditions at a point in the frozen flow solution. Called by FRØZEQ.
PRTA	Organizes the output of conditions at each point in the nonequilibrium flow solution. Called by NØNEQ.
PUTQIN	Organizes the calculations of the cross sections used in computing the transport properties. Called by TRANSP.
QCØUL	Computes the factor $0.8 Q_c$ in the Coulomb cross section for use in the transport property calculations. Called by PUTQIN.
QEX	Computes the cross sections for exchange interaction. Called by PUTQIN.
QEXP	Computes the cross sections for the exponential potential $\phi = Ae^{-r/\rho}$. Called by PUTQIN.
QINTRP	Interpolates tabular data in cross section calculations. Called by QEXP, QLJ, and QTAB.
QLJ	Computes the cross sections for the Lennard-Jones (6-12) potential. Called by PUTQIN.

QMIX Computes transport cross sections from the mixing rule, equation I(102).* Called by PUTQIN.

QREPP Computes the cross sections for an attractive or repulsive power law potential $\phi = Ar^{-\eta}$. Called by PUTQIN.

QSAME Sets the cross sections for a species pair equal to a constant multiple of those computed previously for another pair. Called by PUTQIN.

QTAB Computes the transport cross sections by linear interpolation in tables of $\bar{\Omega}(1,1)$, $\bar{\Omega}(2,2)$, and B^* versus temperature. Called by PUTQIN.

Q11 Computes the transport cross sections using the fairing option, equations I(107) and I(108). Called by PUTQIN.

Q12 Computes the cross sections by the generalized mixing rule, equation I(109). Called by PUTQIN.

Q13 Sets one of the cross sections for a given species pair equal to a constant times another cross section for the same pair. Called by PUTQIN.

Q14 Recomputes the cross sections for a species pair using equations I(112). Called by PUTQIN and Q11.

RADIUS Computes the profile ordinate, the geometric area ratio, and another quantity required in the calculation of the boundary layer at the first point in the flow solution. Called by BLAYER.

*Equations in Volume I (ref. 1) are cited in the form I().

READ	Reads the input data for each case and sets up arrays defining the flow geometry and gas model. Called by MAIN.
RESET	In the IBM 360 version, simulates a UNIVAC 1108 library routine for storing the time at the start of a run. Called by ELTIME.
RESTMP	Computes the equilibrium conditions in the upstream reservoir based on pressure and mass flow or mass flow and stagnation enthalpy data. Called by MAIN.
RNKT	Computes the changes in the species concentrations and other flow variables over a step in the nonequilibrium integration. Called by NØNEQ.
SHØCK	Computes the pressure ratio for an oblique shock in a perfect gas. Called by WEDGE.
SIMQ	Solves systems of linear inhomogeneous equations. Called by DSMSØL and RESTMP.
STUNTS	Prints an edit of species thermal properties. Called by MAIN.
(STUNT2)	Entry point in subroutine STUNTS; prints edits of the steps in the transport cross section calculation. Called by MAIN.
THERM	Computes the species thermal properties at a specified temperature. Called by DERIVS, EQCALC, FRØZEQ, MØDEL, NEWRAP, and STUNTS.
(THERM1)	Entry point in subroutine THERM; computes only the species specific heats at a specified temperature. Called by MAIN, ØUT1, and TRANSP.
.THRØAT	Implements the switch from the upstream solution by the inverse method to the downstream direct integration. Called by NØNEQ.

(TIME) Entry point in subroutine RESET (IBM 360 version only); simulates a library timing routine available on the UNIVAC 1108. Called by ELTIME.

TRANSP Organizes the transport property calculations. Called by BLAYER, FRØZEO, MAIN, MØDEL, ØUT1, WEDGE, and STUNT2.

(TRANSX) Entry point in subroutine TRANSP; sets up arrays used in the transport property calculations. Called by MAIN.

WEDGE Computes and prints the conditions on the surface of a blunt wedge model inserted into the flow. Called by MØDEL.

WESØLN Calculates the quantities z , zz' , and $(zz')'$, based on the approximate solution I(484) of the wedge equation, for given values of ζ and Γ . Called by WEDGE.

XSECT Sets arrays required for applying the precoded cross section data to the species present in the current gas model and sets up the edited cross section calculation. Called by TRANSX.

2.3 UNIVAC 1108 Version

The UNIVAC version of NATA has been set up to run on the 1108 processors at NASA/JSC under the EXEC II operating system. This version is a single precision program in which some arrays and three subroutines (SHØCK, SIMQ, and WESØLN) are typed as double precision to maintain adequate accuracy. These three routines all perform operations which, under some conditions, can lead to loss of many significant digits because of cancellation of terms. The typing of these routines as double precision is accomplished by retaining the IMPLICIT REAL*8 (A-H, Ø-Z) statements from the IBM 360 version. The UNIVAC Fortran V compiler interprets these statements correctly, a feature included by Sperry Rand in the interest of compatibility with IBM 360 Fortran IV.

NATA is too large to be run on the 1108 without use of overlay. Figure 1 shows the overlay structure employed. The overlay is implemented by means of the EXEC II Memory Allocation Processor (ref. 4). In figure 1, the names enclosed in boxes represent the routines of NATA listed in Section 2.2. Each box represents a segment or subsegment with the associated name (S.---) given in parentheses. The level 0 segment S.MAIN is permanently in core. The various level 1 segments overlay one another. In a typical NATA case, the loading sequence for the level 1 segments is as follows:

S.READ	Read input data, set up gas model, etc.
S.LIST	Print problem summary.
S.INIT	Compute reservoir conditions, etc.
S.XSEC	Set up transport cross section calculation for current gas model.
S.ØUT	Print definitions of output labels.
S.FRØZ	Calculate equilibrium solution to throat, to obtain displacement thickness at throat.
S.INIT	Recalculate reservoir conditions corrected for displacement thickness at throat.
S.FRØZ	Calculate frozen solution; calculate equilibrium solution.
S.NØNE	Calculate nonequilibrium solution.

In the nonequilibrium solution, the subsegment S.N2 is loaded first for use in the initial solution by the perturbation method. When the nonequilibrium integration is started, the integration routine RNKT is loaded, overlaying S.N2.

Control cards for using NATA on the UNIVAC 1108 processors at NASA/JSC are shown in figures 2 and 3. Figure 2

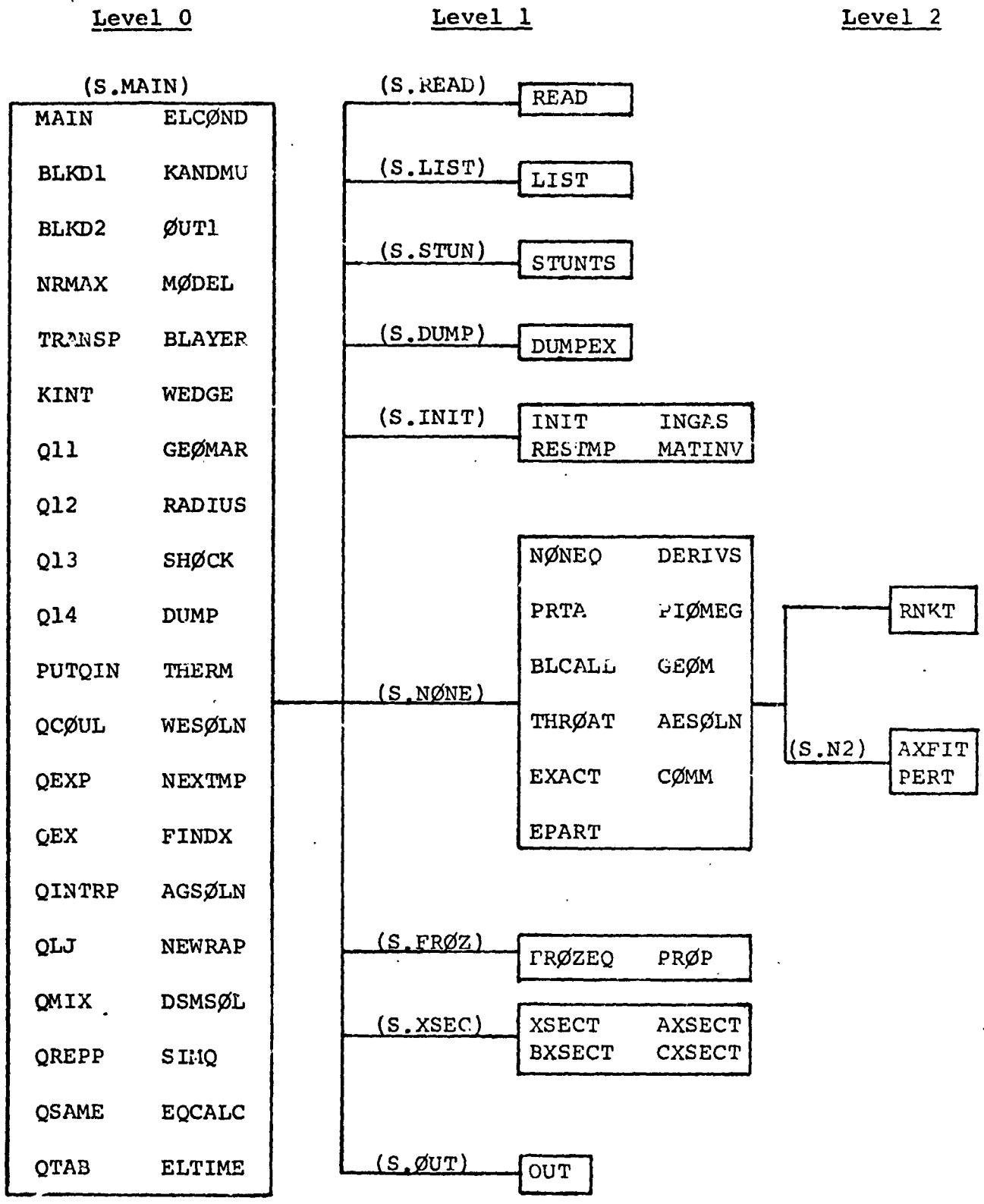


Figure 1. Overlay Structure for UNIVAC 1108 Version of NATA

Z RUN N0320.....

SCOTT/

N MSG FILE REQ. TAPE 01 FH432 0 FSTRN 00

C=OUT

FOR MAIN,MAIN

FOR SUB,SUB

MAP NATA/NATA

S.0 SEG S.MAIN-*(S.READ,S.INIT,S.XSEC,S.FROZ,S.NONE,S.OUT,S.LIST,
S.STUN,S.DUMP)

S.MAIN SEG MAIN-BLKD1-BLKD2-NRMAX-TRANSP-KINT-Q11-Q12-Q13-Q14-PUTGIN-
QCOU-QEXP-QEX-QINTRP-QLJ-QMIX-OREPP-QSAME-QTAB-FLCOND-KANDMU-
OUT1-MODEL-PLAYER-WEDGE-GEOMAR-RADIUS-SHOCK-DUMP-THERM-WESGLN-
NEXTMP-FINDX-A' .OLN-NEWRAP-DSMSOL-SIMQ-EQCALC-ELTIME

S.READ SEG READ

S.LIST SEG LIST

S.STUN SEG STUNTS

S.DUMP SEG DUMPEX

S.INIT SEG INIT-RESTMP-INGAS-MATINV

S.XSEC SEG XSECT-AXSECT-BXSECT-CXSECT

S.FROZ SEG FROZEQ-PROP

S.NONE SEG NONEQ-PRTA-BLCALL-THROAT-EXACT-EPART-DERIVS-PIOMEG-GEOM-AESOLN-
COMM-*(RNKT,S.N2)

S.N2 SEG AXFIT-PERT

S.OUT SEG OUT

XQT CUR

TRW C

JT C

EF C

REL C

Figure 2. Control Cards for Generating a NATA PCF
Tape on the UNIVAC 1108

Z RUN N0320.....

SCOTT/

N MSG FILE REQ. TAPE 01 FH432 0 FSTRN 00

ASG A=V01234

XQT CUR

TRW A

IN A

REL A

XQT NATA

Figure 3. Control Cards for Running NATA on the
UNIVAC 1108

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illustrates a control card deck for generating a NATA PCF tape. The first card is a schematic run card. The deck is set up to compile MAIN and a subroutine "SUB"; the source decks are assumed to be inserted following the "FØR" cards. The "RUN", "MSG", "ASG", "FØR", "MAP", "XQT", "TRW", "ØUT", "TEF", and "REL" cards all have a 7-8 punch in column 1 which was not reproduced by the equipment used to print this listing.

Figure 3 shows the control cards for a NATA run from an existing PCF. The "V01234" following the equals sign on the ASG card is a representative reel number for the PCF tape. The "XQT NATA" card should be followed by the input data for the case, which are described in Section 2 of Volume II (ref. 2).

The control cards required for use of NATA at other UNIVAC 1108 installations may differ in some respects from those illustrated in figures 2 and 3.

2.4 IBM 360 Version

Since NATA is run as a single precision program on the UNIVAC 1108, single precision function names (SQRT, ABS, AMAX1, etc.) are used throughout the program, except in the three double precision routines, SHØCK, SIMQ, and WESØLN. To avoid the tedium of changing function names when the code is converted for use as a double precision program on the IBM 360, the single precision names are retained in the source program for the IBM version and are interpreted as referring to the double precision functions. In the IBM 360/75 installation at Avco Systems Division, this reinterpretation of the single precision Fortran function names is accomplished by using a special subroutine library in the link edit step. At other installations where such a special library may not be available, the same result can be achieved by including, in the NATA source deck, Fortran subroutines defining all of the single precision Fortran function names in terms of the corresponding double precision functions. An example of this procedure is provided by the function EXP in the IBM version of NATA. As shown by the listing of EXP in Section 5, this subroutine redefines EXP in terms of DEXP. It also

sets EXP(X) to 0 for X less than -180, to avoid the production of paper-wasting underflow messages.

Since the Fortran G compiler gives a level 8 diagnostic when a single precision function name is used with a double precision argument, the G compiler cannot be used on the NATA version presented here. Instead, the H compiler, which gives only a level 4 diagnostic, must be used. Several of the NATA routines, including READ and the two block data routines, are too large for the standard H compiler. At Avco Systems Division, an available large compiler (FØRTHMAX) is used to process the NATA source deck.

In both versions of NATA, the elapsed time since the beginning of the run is printed out at several stages in each case. In the UNIVAC version, these times are obtained by calling routines RESET and TIME, which are available in the 1108 system library. In the IBM 360 version in operation at Avco Systems Division, RESET and TIME are simulated using the system library routine ACUCPU, which gives the unexpended CPU time remaining before automatic job termination at TIME.GØ.

In summary, conversion of the UNIVAC version of NATA into the IBM 360 version requires the following operations:

- (1) An IMPLICIT REAL*8 (A-H, Ø-Z) card must be inserted into each NATA routine except ELTIME, ØUT, RESET, SHØCK, SIMØ, and WESØLN. In MAIN, this must be the first card apart from comment cards. In the other routines, the IMPLICIT statement must directly follow the SUBRØUTINE, FUNCTION, or BLØCK DATA statement.
- (2) The subroutines EXP and RESET must be inserted.
- (3) If a special system library for inter, ceting the single precision Fortran function names as double precision functions is not available at the computer installation, Fortran subroutines to define the single precision names as double precision functions must be added to the source deck.

Figure 4 illustrates the control cards used to run NATA from a tape containing the binary deck, on the Avco Systems Division IBM 360/75 system. (The required JCL cards will differ somewhat at other IBM 360 installations). The fourth, fifth, and sixth cards in figure 4 invoke the special system library which interprets the single precision function names as referring to the corresponding double precision functions. The binary tape containing NATA is identified on the DD card with VOLUME = SER = NATA3. The data set GØ.FT08F001 is the binary tape upon which NATA can write data for subsequent plotting. GØ.FT07F001 is the data set for punched output.

```

//          JOB                                .X
//          MSGLEVEL=1
// XEC FORTHLG,REGION.GO=4 20K,TIME.GO=4
//LKED.SYSLIB DD DSNAME=VER1,TEMPLIB,DISP=SHR
// DD DSNAME=SYS1,FORTLIB,DISP=SHR
// DD DSNAME=SYS1,DOUBLEP,DISP=SHR
//LKED.SYSUT1 DD SPACE=(1024,(600,50))
//LKED.SYSLMOD DD SPACE=(3072,(200,10,1),RLSE)
//LKED.SYSLIN DD UNIT=(TAPE9.,DEFER),LABEL=(,BLP),DISP=(OLD,KEEP),    X
//          DCB=(RECFM=FB,LRECL=80,BLKSIZE=3200),VOLUME=SER=NATA3
// DD *
  ENTRY MAIN
/*
//GO.FT08F001 DD UNIT=(TAPE9.,DEFER),LABEL=(,BLP),DISP=(NEW,KEEP),    C
//          LCB=(RECFM=VBS,LRECL=92,BLKSIZE=4604),VOLUME=SER=DATASV
//GO.FT07F001 DD SYSOUT=B
//GO.FT05F001 DD *

```

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Figure 4. Control Cards for Running NATA on the IBM 360

3. ANALYSIS OF ROUTINES

The present section discusses and explains the main program and each subroutine of the NATA code. The purpose of this exposition is to provide an entry into the coding for programmers who wish to analyze errors, make corrections, or introduce modifications into NATA.

All of the physics and gas dynamics and most of the mathematical analysis underlying the programming have already been documented in Volumes I and II of this report (refs. 1, 2). Where appropriate, individual sections and equations of Volumes I and II are referenced.* In cases where the underlying mathematical analysis has not been documented previously, it is explained in the present section.

The routines of NATA are discussed in the following order: first the main program, then all subroutines and functions in alphanumeric order of their names. This is the same order as that in which the routine names are listed in Section 2.2 and at the beginning of Section 5, and in which the source listings are presented in Section 6.

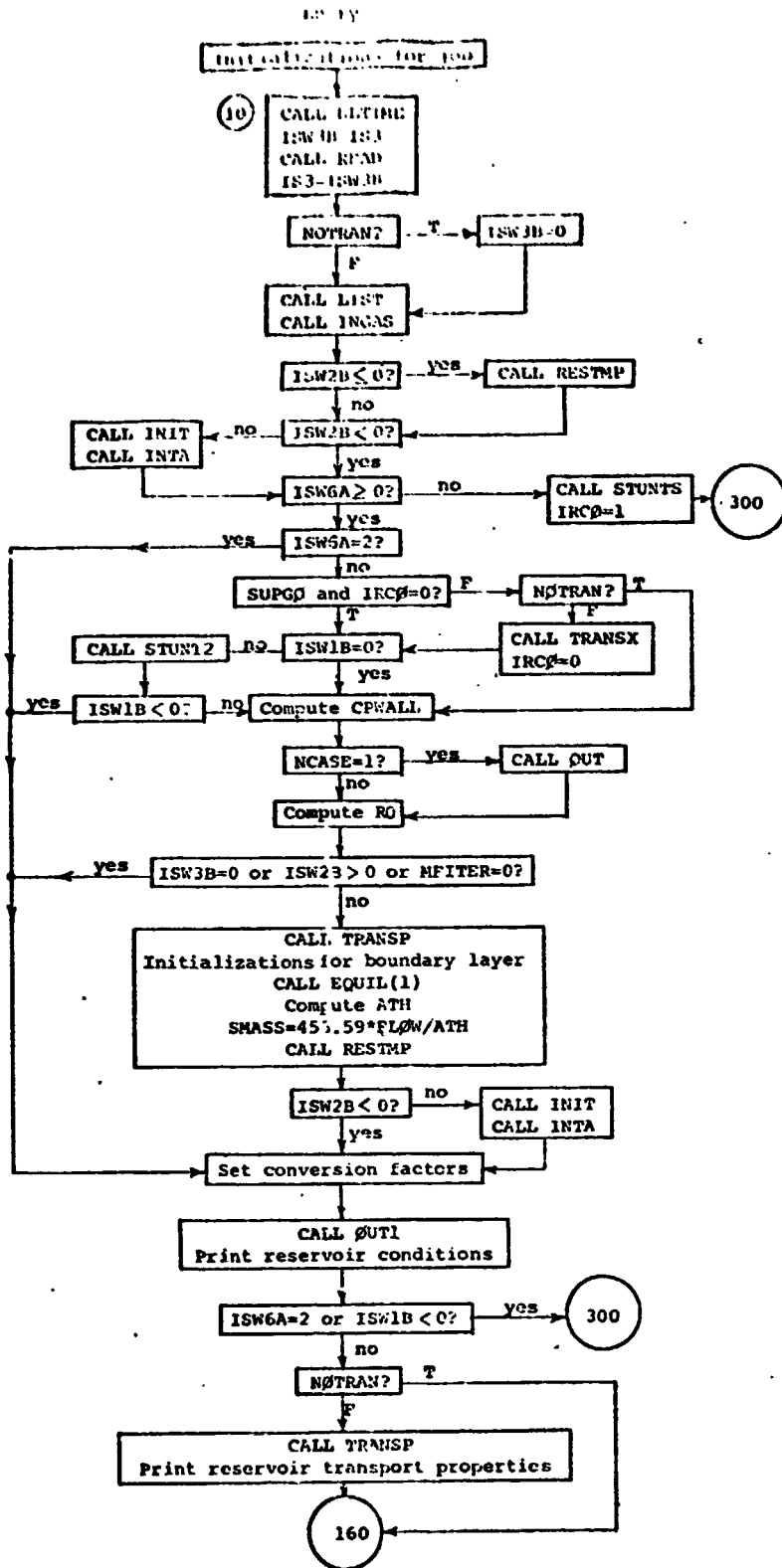
3.1 Main Program

The main program performs certain initialization functions, calls the major subroutines in proper sequence, implements some of the input options, and controls job termination. A flowchart of MAIN is shown in figure 5. The operations shown in this flowchart will be discussed approximately in the order in which they are performed during execution of the code.

3.1.1 Job Initialization

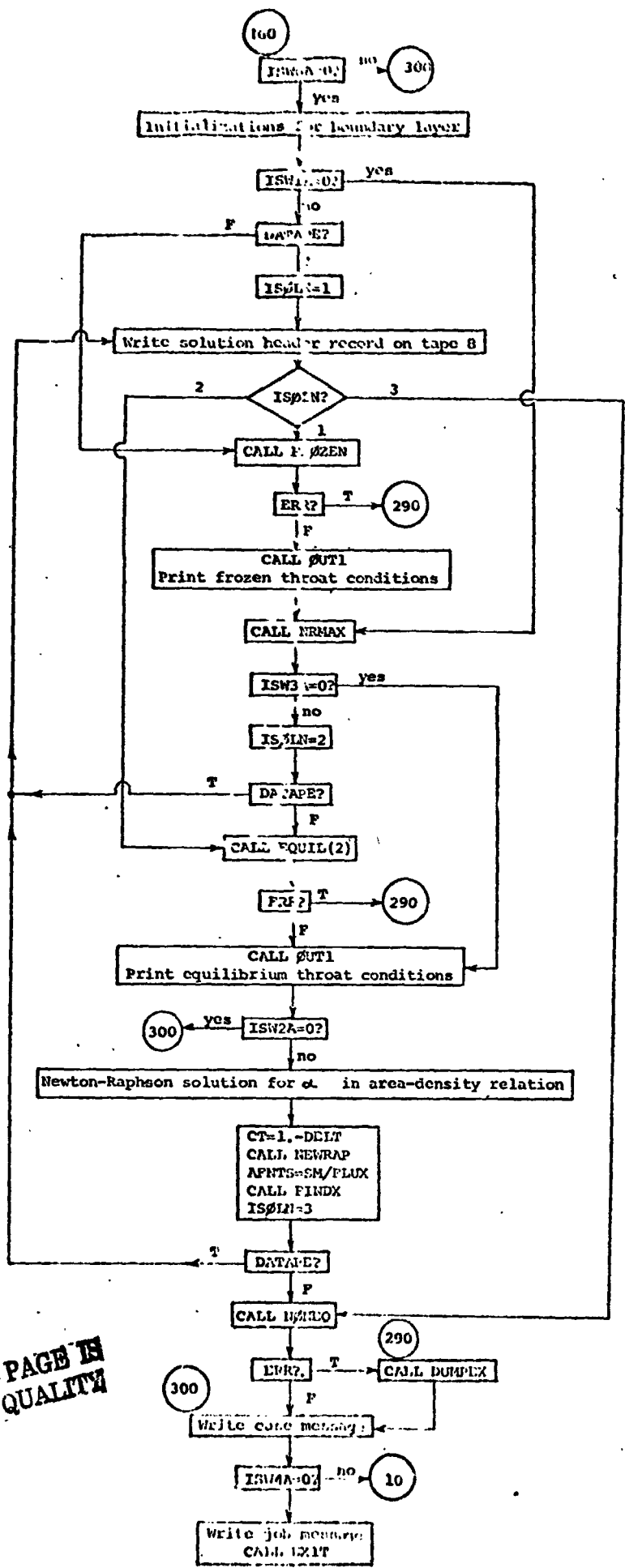
The 12 input control parameters ISW1A to ISW6B are initialized by arithmetic statements in MAIN before any input data

*Equations from Volumes I and II will be referenced in the form I() and II().



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Figure 5a. Flowchart of NATA Main Program (Part a)



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have been read. This procedure is used because these parameters are in unlabelled common, and hence cannot be preset in a BLOCK DATA routine.

3.1.2 Case Initialization

The calculations for each case in a job always begin with statement 10. First, subroutine ELTIME is called to determine the elapsed time ETO since the start of the first case. Then subroutine READ is called. This routine reads the input data for the case, sets up arrays describing the gas model and the nozzle geometry, and prints some of the data defining the case being run. Next, subroutine LIST is called to print the remaining data defining the current case. Subroutine ELTIME is then called again to print the time. ELTIME is called at various locations in MAIN, to provide data on the execution time consumed by major portions of the calculation. These subsequent calls to ELTIME are omitted from the flowchart in the interest of simplification.

Before the call to LIST, the value of the logical input variable NØTRAN is checked. NØTRAN is a switch to allow the suppression of all transport property calculations throughout NATA. With NØTRAN = .TRUE., flow solutions can be run for a gas model containing user-specified species for which transport cross section data have not been provided. Of course, when the transport property calculations are suppressed, the boundary layer cannot be included in the flow solutions. Hence, for NØTRAN = .TRUE., ISW3B is set to zero. However, the input value of ISW3B is saved in the location IS3, and is reset before the call to READ in subsequent cases.

MAIN next increments the case counter NCASE and initializes some variables including the indicator ERR in common block /ERRØR/. This indicator is set to .FALSE. near the beginning of each case. If the programmed checks in any subroutine subsequently detect an error, ERR is set to .TRUE., and this indication causes control to be returned to MAIN as quickly as possible, with no further calculations. Statements in MAIN then cause subroutine DUMPEX to be called. DUMPEX prints out a diagnostic dump consisting of the current values of most of the variables in labelled and unlabelled

common. The details of this scheme are explained more fully in the sections (below) on subroutines DUMP and DUMPEX. Most of the tests on ERR in MAIN have been omitted from the flow-chart for brevity, but the ones following CALL FRØZEN, CALL EQUIL(2), and CALL NØNEQ have been included to illustrate the method.

MAIN next calls subroutine INGAS to set up certain arrays used to describe the chemical nature of the gas mixture. Then, if $ISW2B \leq 0$ (i.e., if the determination of reservoir conditions is to be based partly on the total mass flow), subroutine RESTMP is called to compute the reservoir temperature and pressure. Unless ISW2B is negative, subroutine INIT and INTA are then called. INIT initializes a number of control variables and nondimensionalizes some species properties using the reservoir temperature. INTA computes the species mole fractions and the thermal properties of the gas in the reservoir. For negative ISW2B, INIT and INTA are called in RESTMP with the final reservoir temperature and pressure, and thus do not have to be called in MAIN.

3.1.3 Special Options

If ISW6A is negative, NATA computes and prints tables of thermal properties for all of the species in the current gas model, instead of calculating a flow solution. This operation is performed by subroutine STUNTS.

If ISW6A is positive, NATA computes and prints out the reservoir conditions (as usual), but does not proceed with the flow solutions. For the specific value $ISW6A = 2$, the calculation of the transport properties in the reservoir is also omitted. This feature allows the NATA user to employ the code for thermochemical calculations for a gas model containing nonstandard species without setting up transport cross section data for the species.

If ISW6A is 0 (the normal, preset value), NATA generates one or more flow solutions.

If ISW6A is zero or positive (but not equal to 2), the transport properties in the reservoir are computed. Before transport property calculations can be done, the entry TRANSX of subroutine TRANSP must be called. However, the call to TRANSX can be omitted if TRANSX has been called in a previous case with the same gas model. The gas model is the same as in the preceding case if SUPGØ = .TRUE., and TRANSX was called in a previous case if IRCØ = 0. If NØTRAN is .TRUE., the call to TRANSX is skipped.

Another special option is an edit of the transport cross section calculations, performed by calling the entry STUNT2 of subroutine STUNTS. This call is executed if the input control parameter ISW1B is nonzero. For ISW1B negative, STUNT2 also prints a table of averaged transport cross sections; in this case, no flow solution is run.

3.1.4 Specific Heat of the Gas at the Wall Temperature

MAIN computes the specific heat of the gas at the nozzle wall temperature, TWALL. This calculation is based on the assumption that the gas at this low temperature is a mixture of the "cold species" used to define the elemental composition. The specific heat is calculated as

$$C_{pw} = \frac{R_0}{W_c} \sum_{k=1}^{n_c} x_k \frac{C_{pk}}{R_0} \quad (1)$$

in which W_c is the mean molecular weight of the cold gas, n_c the number of cold species, x_k the mole fraction of the k^{th} cold species, and C_{pk}/R_0 the nondimensional molar specific heat at constant pressure for the k^{th} cold species. This species specific heat is given by CCPJ(I), where $I = IJCS(K)$ is the index, in the list of species for the current gas model, of the k^{th} cold species.

3.1.5 Definitions of Output Variable Labels

In the first case of each job (NCASE = 1), MAIN calls subroutine ØUT to print a list of definitions, with units,

of the output variable labels used in printing out the flow conditions and test conditions on models. These labels differ, in most cases, from the Fortran variable names used in the code.

3.1.6 Correction of the Reservoir Conditions for the Displacement Thickness at the Throat

If the boundary layer is to be included in the solution ($ISW3B \neq 0$), and if the reservoir condition calculations are based partly on the total mass flow ($ISW2B \leq 0$), then NATA normally recalculates the reservoir conditions, taking account of the effects of the boundary layer displacement thickness on the equilibrium sonic mass flux. For $ISW2B \leq 0$, subroutine RESTMP determines the reservoir temperature and pressure so as to match the sonic mass flux SMASS and one other condition (on pressure or enthalpy). Initially, when RESTMP is called the first time, SMASS is calculated as the ratio of the input total mass flow to the geometric cross sectional area of the nozzle at the throat. To correct these results for the displacement thickness effect, MAIN calls entry EQUIL(IPASS) of subroutine FRØZEQ with the argument IPASS = 1. EQUIL then computes the equilibrium solution up to the throat, including the boundary layer. MAIN then calculates the effective cross sectional area ATH at the throat including the effects of the displacement thickness, recomputes the sonic flux SMASS, and calls RESTMP for a second time to recompute the reservoir conditions. This correction can be suppressed, if it is not desired, by setting MFITER = 0 in the input.

3.1.7 Conversion Factors

Next, MAIN sets the array of conversion factors CF(I) in common block /CØNVRT/. These factors convert the non-dimensional flow variables used in the internal computations to the units used in the printed output.

3.1.8 Output of Reservoir Conditions

The nondimensional flow variables $CT = T/T_0$, $PRES = p/p_0$, etc., are now set to their reservoir values and subroutine ØUT1 is called to print out the reservoir conditions.

3.1.9 Solution Header Records on Tape 8

If the frozen flow solution is to be generated ($ISW1A \neq 0$) and if data are to be written on a binary output tape for subsequent plotting ($DATAPE = .TRUE.$), the main program next writes a "header" record on tape 8 for the frozen solution. This record contains the nozzle index $N\emptysetZZLE = NPR\emptysetFL(1)$, the type of solution ($IS\emptysetLN = 1$ for frozen flow), and the reservoir conditions.

3.1.10 Frozen Solution

To generate the frozen flow solution, MAIN calls the entry $FR\emptysetZEN$ of subroutine $FR\emptysetZEQ$. It then calls $\emptysetUT1$ to print the sonic-point conditions in the frozen solution.

3.1.11 Equilibrium Solution

MAIN calls subroutine $NRMAX$ to compute the conditions at the sonic point in the equilibrium solution. These data are needed for both the equilibrium and nonequilibrium solutions. Then, unless $ISW3A = 0$, MAIN checks $DATAPE$. If this control variable is $.TRUE.$, a header record is written on tape 8 for the equilibrium solution, and the entry $EQUIL(IPASS)$ of subroutine $FR\emptysetZEQ$ is called with $IPASS = 2$ to produce the solution. Finally, $\emptysetUT1$ is called to print out the equilibrium throat conditions.

3.1.12 Nonequilibrium Solution

If the nonequilibrium solution is to be computed ($ISW2A \neq 0$), MAIN determines the constants C and α in the area-density relation $I(383)$ assumed in the solution by the inverse method upstream of the throat. The parameter α is determined by a Newton-Raphson solution of equation $I(385)$ or $I(386)$. The iteration is started using the initial estimate

$$\alpha \approx 4 + 8 \ln \left(\frac{\rho}{\rho_0} \right) \quad (2)$$

which can be derived by taking the logarithm of $I(385)$ and

expanding $\ln(\alpha + 2)$ around $\alpha = 0$:

$$\ln(\alpha + 2) \approx 2 + \frac{\alpha}{2} - \frac{\alpha^2}{8} \quad (3)$$

Then to obtain starting conditions for the perturbation solution, MAIN sets the nondimensional temperature CT to 1-DELTA, calls subroutine NEWRAP to compute the equilibrium flow conditions at this temperature, computes the corresponding effective area ratio AFNTS, and calls subroutine FINDX to determine the axial coordinate CX at which these conditions occur. If DATAPE = .TRUE., a header record is written on tape 8 for the nonequilibrium solution. Then subroutine NØNEQ is called to generate the nonequilibrium flow solution.

3.1.13 Diagnostic Dumps

If a .TRUE. value of the error indicator ERR is detected anywhere in MAIN, control is immediately transferred to the statement CALL DUMPEX, which prints out the current values of a large number of common variables as an aid to identifying the cause of failure. If data have been written on tape 8, the tape is backspaced to eliminate the records written during the failed case.

3.1.14 Job Termination

At the end of the calculations for each case, a line of output is written containing the case number, the number of records written on tape 8 (if any), and the total number of records written thus far in the job. Then the value of the job termination control parameter ISW4A is checked. If ISW4A is nonzero, the input data for the next case are read. If ISW4A is zero, a summary of the completed and failed cases in the job is written and, if DATAPE is .TRUE. an end file mark is written on tape 8 and the tape is rewound.

3.2 Subroutine AESØLN

This subroutine calculates the effective area ratio at a given position, X, in the nozzle. The value X is a subroutine

argument. Other input data to the routine are contained in the common block /AEGEØM/, namely, the geometric area ratio S1, the derivative of the geometric area ratio S2, and the square root of the geometric area ratio SQRTA. To set S1 and S2, a call to AESØLN(X) must be preceded by a call to GEØMAR(X,S1,S2). In addition, if NPRFLS = 1 (the flow is in a nozzle, not a channel) and JDIM = 1 (the nozzle is axisymmetric), then SQRTA must be computed in the calling routine before AESØLN is called.

The calculations of the effective area ratio are based upon equations I(126), I(130) and I(134), respectively, for the cases of a two-dimensional nozzle (NPRFLS = 1, JDIM = 0), an axisymmetric nozzle (NPRFLS = 1, JDIM = 1), and a rectangular channel (NPRFLS = 2). The effective area ratio obtained is stored in the location S1 of common block /AEGEØM/, replacing the geometric area ratio.

In addition, the subroutine computes the derivative, dA_e/dx , of the effective area ratio, using the following formulas:

2D Nozzle

$$\frac{dA_e}{dx} = \frac{\frac{dA_g}{dx} - \frac{1}{y_0} \frac{d\delta^*}{dx}}{1 - \delta^* / y_0} \quad (4)$$

Axisymmetric Nozzle

$$\frac{dA_e}{dx} = \sqrt{A_e} \left[\frac{\frac{1}{\sqrt{A_g}} \frac{dA_g}{dx} - \frac{2}{y_0} \frac{d\delta^*}{dx}}{1 - \delta^* / y_0} \right] \quad (5)$$

Channel

$$\frac{dA_e}{dx} = \frac{(z - \delta_2^*) \left(\frac{dy}{dx} - \frac{d\delta_1^*}{dx} \right) + (y - \delta_1^*) \left(\frac{dz}{dx} - \frac{d\delta_2^*}{dx} \right)}{(y_0 - \delta_{1*}^*) (z_0 - \delta_{2*}^*)} \quad (6)$$

These relations can easily be derived from equations I(126), I(130), and I(134). The computed value of dA_e/dx is stored in the location S2 of common /AEGEØM/, where it replaces the derivative of the geometric area ratio.

3.3 Subroutine AGSØLN

The purpose of this subroutine is to solve for the geometric area ratio A_g when the effective area ratio A_e and boundary layer displacement thickness δ^* are given. These input data are provided in the argument list: $A_e = AE$, $\delta_i^* = DEL(i)$, where the index i runs from 1 to 2 in the case of a channel. The argument list also includes the indicator UPDØWN, which specifies whether an upstream (-1.) or downstream (+1.) solution is desired. The computed value of A_g , and the corresponding value of x , are returned to the calling routine through the arguments AG and X, respectively.

The solution for A_g is straightforward and explicit in the case of a nozzle. It is given by equation I(127) if the flow geometry is two-dimensional and by I(131) if it is axisymmetric. In both cases, the value of X corresponding to A_g is determined by calling FINDX.

In the case of a channel, the relation between A_e and A_g is given by equation I(135) and the profile functions $y(x)$, $z(x)$. This relation does not permit an explicit solution for A_g . In AGSØLN, a numerical Newton-Raphson method is used to determine A_g . Equation I(135) is written as

$$A_g = A_c + \frac{y \delta_2^* + z \delta_1^*}{y_0 z_0} \quad (7)$$

where

$$A_c = \left(1 - \frac{\delta_1^*}{y_0}\right) \left(1 - \frac{\delta_2^*}{z_0}\right) A_e - \frac{\delta_1^* \delta_2^*}{y_0 z_0} \quad (8)$$

The quantity A_c and the factors δ_1^*/y_0z_0 and δ_2^*/y_0z_0 are constants during the solution for a particular call to the subroutine. From (7), the object of the calculation is to find a value of x such that $F(x) = 0$, where

$$F(x) \equiv A_c + \frac{y(x) \delta_2^* + z(x) \delta_1^*}{y_0z_0} - A_g(x) \quad (9)$$

Let x, F denote the values of these quantities from the current iteration and x_0, F_0 the values for the previous iteration. Then, based on a linear approximation to (9), the value of x required to make $F(x) = 0$ is estimated to be

$$x_n = x - F \cdot \frac{x - x_0}{F - F_0} \quad (10)$$

The iteration based on this relation begins at statement 60 in the subroutine. To start the iteration, A_g is set equal to A_e , and the corresponding x is computed using FINDX. The second point is obtained by computing A_g from (7) and again using FINDX. Equation (10) is then used in the third and all subsequent steps. The convergence criterion is that $|F|/A_g$ be less than or equal to 10^{-5} , or that $F = F_0$.

If $A_g(x)$ in (9) is expressed by means of equation I(120), equation (9) can be rewritten in the form

$$F(x) = \left(1 - \frac{\delta_1^*}{y_0}\right) \left(1 - \frac{\delta_2^*}{z_0}\right) A_e - \left(\frac{y}{y_0} - \frac{\delta_1^*}{y_0}\right) \left(\frac{z}{z_0} - \frac{\delta_2^*}{z_0}\right) \quad (11)$$

If y and z both have minima at the throat $x = 0$, as is normally the case in channels, then the maximum value of $F(x)$ occurs at $x = 0$ and is equal to

$$F_{\max} = F(0) = \left(1 - \frac{\delta_1^*}{y_0}\right) \left(1 - \frac{\delta_2^*}{z_0}\right) A_e - \left(1 - \frac{\delta_1^*}{y_0}\right) \left(1 - \frac{\delta_2^*}{z_0}\right) \quad (12)$$

In order for a solution of the equation $F(x) = 0$ to exist, it is necessary that F_{\max} be greater than or equal to zero. This is always true if $\delta_1^* = \delta_{1*}^*$ and $\delta_2^* = \delta_{2*}^*$. Also, in practice, it is always true when $A_e \gg 1$. However, when A_e is nearly equal to 1 and $\delta_1^* < \delta_{1*}^*$, $\delta_2^* < \delta_{2*}^*$, F_{\max} can be negative. These circumstances can arise near the sonic point (where $A_e = 1$) when the sonic point is displaced from the geometric throat by the presence of large gradients in the displacement thicknesses. When this occurs, the iteration formula (10) extrapolates past the throat and gives an x_n of the wrong sign, i.e., the opposite sign to UPDOWN. When such a condition is detected, AGSØLN attempts to "fix" the problem by resetting the displacement thicknesses δ_1^* , δ_2^* to their values at the sonic point, δ_{1*}^* and δ_{2*}^* respectively, and restarting the iterative solution at the beginning. Under the circumstances in question, this adjustment can have little effect on the value of A_g computed (which is near unity). Equation (12) shows that after this replacement, F_{\max} is the product of a positive factor and $(A_e - 1)$, and hence is ≥ 0 . Thus, if the problem was indeed caused by the combination of circumstances described above, the "fix" should guarantee the existence of a solution of the equation $F(x) = 0$. The coding provides a dump and an error exit should the problem recur after the "fix" has been carried out. In this case, the problem has a different cause, such as an error in a user-supplied channel profile curvefit.

3.4 Subroutine AXFIT

This subroutine solves for the value of x corresponding to a given effective area ratio (AFNTS) during the initial (perturbation) portion of the nonequilibrium solution. If the solution is being calculated without the boundary layer (ISW3B = 0), then the geometric area ratio is equal to the effective area ratio, and x (denoted by CX) is determined by calling subroutine FINDX. If the boundary layer is included in the solution, but subroutine THRØAT has not yet been called to reset IUPD from 1 to 0, the effects of the boundary layer displacement thickness on the effective flow geometry are neglected, and x is again obtained from FINDX. However, if the boundary layer is included and IUPD = 0, then the inviscid flow is coupled to the boundary layer and

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OF POOR QUALITY

x is accordingly determined by calling AGSØLN. Both FINDY and AGSØLN require an indicator which specifies whether the upstream (UPDØWN = -1) or downstream (UPDØWN = 1) solution for the x corresponding to the effective area ratio AFNTS is desired. In AXFIT, this indicator is determined by testing whether the nondimensional temperature CT is greater than or less than its value CTMAX at the geometric throat based on the equilibrium flow solution.

3.5 Subroutine AXSECT

Subroutine AXSECT is called by subroutine XSECT for the purpose of adding data for additional species pairs to the edited cross section data in common block/TRANS7/. The first argument LL of the subroutine gives the step in the cross section computations at which the additional pairs are to be added, while the second argument NN gives the total number of species to be considered in adding the data. The indices of the actual species to be considered in the added data are then given, in numerical order, by the first NN entries of the variable I in common block/TRANS2/. Subroutine AXSECT searches through all pairs (II, JJ) of these species with $II \leq JJ$, to find those for which cross section data have not been previously specified, as indicated by the fact that $Q(1, II, JJ) = 0$. All such pairs are then added to the IQ, JQ arrays, starting at the location $NQ(LL) + 1$, with the original data in these locations being moved back to make room for the new values. Further, the subroutine sets $Q(1, II, JJ) = 1$ for all pairs added to the IQ, JQ arrays and revises the NQ array to be consistent with the new IQ, JQ values, assuming that all of the added pairs are to be included in the LLth step of the cross section computations.

3.6 Subroutine BLAYER

The boundary layer calculations in NATA are carried out by the method described in Section 5 of Volume I (ref. 1). These calculations are performed by subroutine BLAYER. The boundary layer properties are calculated from the integral I, equation I(172). This integral is used to determine the correlation parameter n, equation I(174). The momentum thickness

θ , displacement thickness δ^* , shear stress τ_w , and heat flux q_w are then calculated from n and other quantities.

The overall structure of BLAYER is diagrammed in figure 6. The calculations are all contained in a $D\emptyset$ loop with index L running from 1 to NPRFLS. In the case of a nozzle (NPRFLS=1), the calculations are executed only once. For a channel (NPRFLS=2), they are executed twice to compute the boundary layers on the two pairs of channel walls.

The subroutine argument FINAL is a logical variable specifying whether the new flow point is an actual point of the solution or merely an intermediate step in the Runge-Kutta integration used to generate the nonequilibrium flow solution. For FINAL = .FALSE., the important variables used in BLAYER are restored to their values at the beginning of the step before the RETURN is executed, and the calculations of shear stress TAUW and heat flux QWD \emptyset T are omitted.

IP \emptyset INT is a counter for the flow points at which boundary layer calculations are done in each flow solution. IP \emptyset INT is initialized to 1 in subroutine FR \emptyset ZEQ for frozen and equilibrium solutions and in subroutine N \emptyset NEQ for nonequilibrium solutions. Each time BLAYER is called with FINAL = .TRUE., IP \emptyset INT is incremented by 1 before the RETURN.

Some of the physical variables which are used frequently are given short mnemonic names which are equivalenced to their standard names in unlabelled common

3.6.1 Initializations

When BLAYER is called for the first time in a given flow solution (IP \emptyset INT = 1), a number of quantities are set or initialized by the statements down to 90. These quantities include the coefficient B in the linear relation I(164) between the momentum parameter N and the correlation parameter n , and the transport properties $PRW = Pr_w$ and $VISCW = \mu_w$ at the wall temperature. The wall is assumed to be catalytic; hence, these transport property calculations are based on the mole fractions QPJ(K) for the cold gas mixture. This section of the routine also sets the indicator

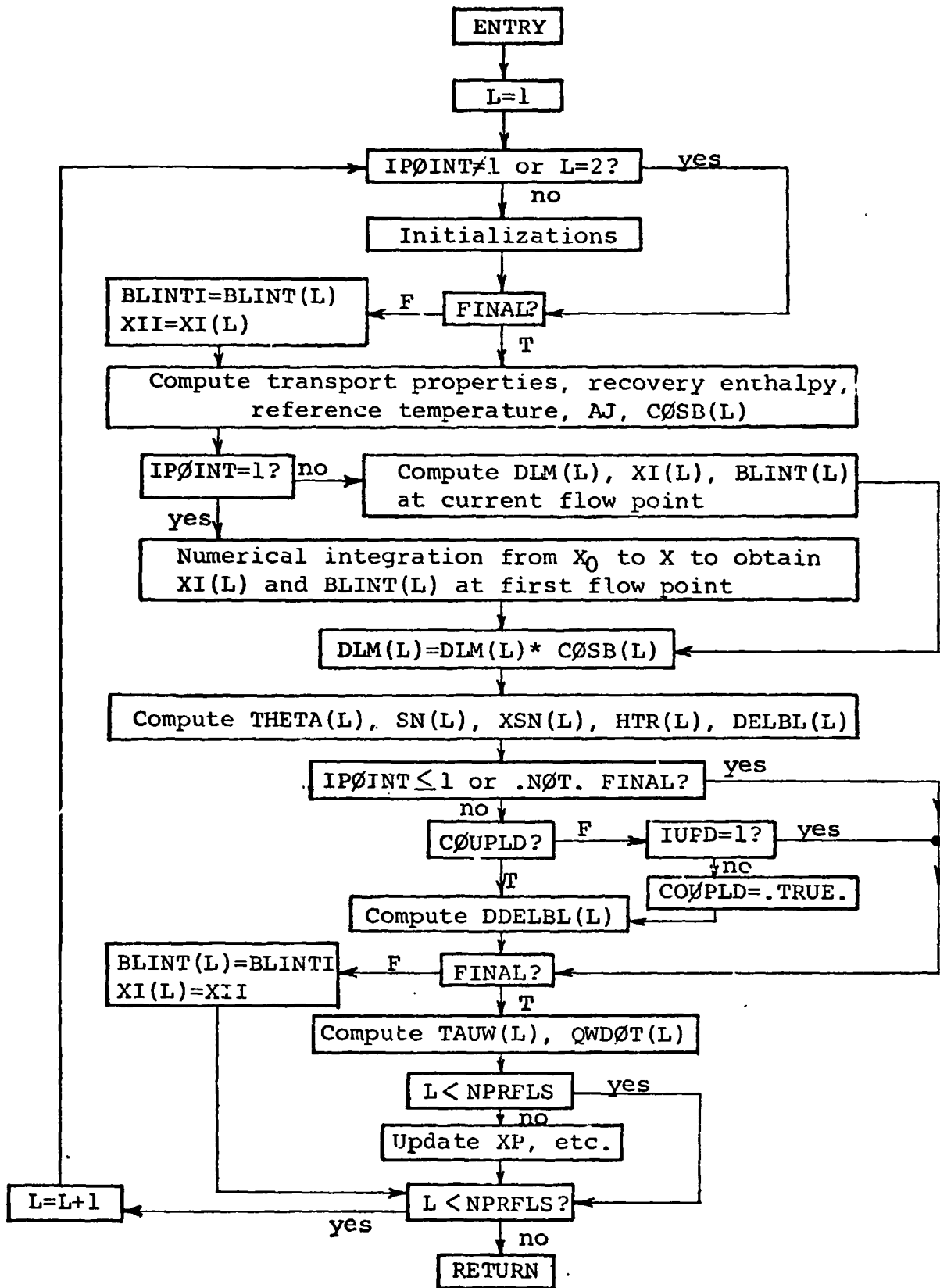


Figure 6. Flow Chart of Subroutine BLAYER

ITYPE, which specifies the type of nozzle geometry:

ITYPE = 1 Two-dimensional nozzle

ITYPE = 2 Axisymmetric nozzle

ITYPE = 3 Rectangular channel

3.6.2 Transport Property and Geometry Calculations

The section of BLAYER from statement 90 down to 270 computes the values of various physical quantities at the current flow point, including the transport properties and geometric parameters entering into the boundary layer calculation.

BLAYER uses the viscosity and Prandtl number at three temperatures, the wall temperature TWALL, the free-stream temperature, TE, and the reference temperature TREF. The calculations for TWALL are done only once per solution because the wall temperature is assumed constant. To save computing time, the transport property calculations at TE and TREF are not repeated at every flow point, but are done only when the temperature or the gas mole fractions have changed significantly. The criterion for performance of the transport property calculations is a change of 0.01 in any of the mole fractions or a change of 0.5 percent in the temperature. The mole fractions at which calculations were last done are represented by XTEST(J). The corresponding temperatures are TEP and TREFP.

The factor r^{2j} in equation I(171) is denoted by AJ in BLAYER. Because of the form of equations I(171) to I(173), r^{2j} can be replaced by a constant times r^{2j} without affecting the computed results for the boundary layer thicknesses, the heat flux, and the shear stress. For two-dimensional nozzles, AJ is set to 1, which is the value of r^{2j} for $j = 0$. For axisymmetric nozzles, AJ is set to the geometric area ratio,

$$A_g = (y/y_0)^2 \quad (13)$$

which is r^{2j}/y_0 . In the case of a rectangular channel, for each profile, AJ is set to the square of the radius of the other profile, since this radius is proportional to the width of the surface whose boundary layer is being calculated.

For nozzles, the cosine of the angle b between the profile tangent and the axis is calculated from dA_g/dx . In general,

$$\cos b = \frac{1}{\sqrt{1 + \tan^2 b}} = \frac{1}{\sqrt{1 + (dy/dx)^2}} \quad (14)$$

where y is the profile ordinate. In the case of a two-dimensional nozzle, from equation I(118),

$$\frac{dy}{dx} = y_0 \frac{dA_g}{dx} \quad (15)$$

and hence

$$\cos b = \frac{1}{\sqrt{1 + y_0^2 (dA_g/dx)^2}} \quad (16)$$

In the axisymmetric case, from I(116),

$$\frac{dy}{dx} = \frac{y_0}{2\sqrt{A_g}} \frac{dA_g}{dx} \quad (17)$$

and thus

$$\cos b = \frac{1}{\sqrt{1 + \frac{y_0^2}{4 A_g} \left(\frac{dA_g}{dx} \right)^2}} \quad (18)$$

The derivative dA_g/dx in (16) and (18) is obtained by calling subroutine GEØMAR. In the case of a channel, dy/dx is computed by calling the entry GMAR3 of GEØMAR, and $\cos b$ is evaluated directly from (14).

The viscosity-temperature exponent ω , defined by equation I(191), is determined by fitting I(191) to the viscosity at the wall and reference temperatures. The coefficient A in I(164) (denoted by EK1) is calculated from I(201) at the first flow point (IPØINT = 1), and then held constant through the solution.

The integrand \mathfrak{F} of I(172) is represented by ØRDIN(L). Actually, ØRDIN(L) differs from equation I(171) by a constant factor, which cancels out in the calculation of n' I(173) and thus has no effect on the boundary layer properties.

3.6.3 Integration from x_0 to First Flow Point

In general, the flow point at which BLAYER is first called lies upstream of the throat. The Mach number at this point is low. However, the point may be well downstream of the point x_0 at which boundary layer development is assumed to start (see Sec. 5.10 of Volume I). Accordingly, the interval from x_0 to the first point of the flow solution is not treated as a single step for the purpose of the boundary layer calculation. Instead, when BLAYER is called with IPØINT = 1, the integral I(172) (denoted by BLINT(L)) and the streamwise boundary layer coordinate ξ (denoted by XI(L)) are evaluated by a 10-step numerical integration from x_0 to the first flow point.

In the region covered by this integration, the Mach number is low so that the flow can be approximated as incompressible. Also, the boundary layer is thin in this region, in comparison with the nozzle or channel radius, so that the effective area ratio can be approximated by the geometric area ratio. Thus, the continuity equation $\rho u A_e' = \text{constant}$, can be approximated

$$M A_g = \text{constant} \quad (19)$$

The constant in (19) is represented by CC. At each point in the integration from x_0 to x_1 , the geometric area ratio $A_g = A_{g1}$ and the quantity A_{gJ} corresponding to r^{2j} are calculated by calling subroutine RADIUS, M is obtained from (19), and a quantity ØRD1 proportional to ØRDIN(L) is computed with the

approximation $a_e/a_0 \approx p_e/p_0 \approx 1$. The increment in ξ is computed using the Pythagorean theorem:

$$(\Delta \xi)^2 = (\Delta x)^2 + (\Delta y)^2 \quad (20)$$

where Δy is the change in the profile ordinate due to a change Δx in the axial coordinate. BLINT(L) is calculated using trapezoidal-rule integration. At the first flow point, DLM(L) = $d \ln M_e / dx$ is evaluated from (19).

3.6.4 Derivatives of the Mach Number

The derivative dM_e/dx of the flow Mach number is an important variable in the boundary layer calculation because it represents the effect of a streamwise pressure gradient. The related quantity $dM_e/d\xi$ appears in the formula I(174) for the correlation parameter, n . All of the boundary layer properties depend upon n .

The derivative $dM_e/d\xi$ differs between the two profiles of a channel because the boundary layer coordinate lies along the channel surface and the two profiles have different inclinations to the nozzle axis. The value of $d \ln M_e / d\xi$ for the L^{th} profile is denoted by DLM(L). It is determined by first computing $d \ln M_e / dx$ and then correcting for the inclination of the nozzle or channel surface to the axis. For each flow point beyond the first, $d \ln M_e / dx$ is calculated from rate data if the flow solution is of non-equilibrium type and the numerical integration technique is being used (INEQ = 1). Otherwise, if the flow solution is of frozen or equilibrium type or if the perturbation technique is being used in a nonequilibrium solution (INEQ = 0), $d \ln M_e / dx$ is calculated by numerical differentiation. If the inviscid solution is being calculated by nonequilibrium integration, DLM(L) is calculated from equations I(420), I(421), and I(291) or I(422). In the case of a chemical nonequilibrium gas model (NT = 1), the stagnation enthalpy h_0 is constant and hence I(421) and I(291) give

$$\frac{du}{dx} = -\frac{1}{u} \sum_{j=1}^n \left[\gamma_j \frac{dH_j}{dx} + H_j \frac{d\gamma_j}{dx} \right] \quad (21)$$

The variables used in the code are

$$SHJ(j) = \frac{H_j}{R_c T_0} \quad (22a)$$

$$CCPJ(j) \cdot DT = \frac{1}{R_0 T_0} C_{pj} \frac{dT}{dx} = \frac{1}{R_0 T_0} \frac{dH_j}{dx} \quad (22b)$$

$$SU = u \sqrt{\frac{W_0}{R_0 T_0}} \quad (22c)$$

$$GJ(j) = \gamma_j \quad (22d)$$

Hence (21) is coded as

$$\begin{aligned} DUDX \equiv \frac{d(SU)}{dx} &= - \frac{W_0}{SU} \sum_{j=1}^n [GJ(j) \cdot CCPJ(j) \cdot DT + SHJ(j) \cdot DGJ(j)] \\ &= - \frac{CMA}{SU} (SCPG \cdot DT + SHJDGJ) \end{aligned} \quad (23)$$

The sums denoted by SCPG and SHJDGJ are obtained from common. Similarly, in the case of an electronic nonequilibrium model (NT = 2), equations I(421) and I(422) give

$$\begin{aligned} DUDX &= \frac{DCHA}{SU} - \frac{CMA}{SU} [GJ(1) \cdot CCPJ(1) \cdot DTE \\ &\quad + SCPGH \cdot DT + SHJDGJ] \end{aligned} \quad (24)$$

where

$$DCHA = \frac{W_0}{R_c T_0} \frac{dh_0}{dx} \quad (25)$$

$$SCPGH = \sum_{j=2}^n GJ(j) \cdot CCPJ(j) \quad (26)$$

In (24), quantities subscripted (1) refer to the electron species.

For equilibrium and frozen solutions, and for nonequilibrium solutions which are being developed by the perturbed equilibrium flow technique (INEQ = 0), the quantities SCPG and SHJDGJ are not available. In these cases, DLM(L) is calculated by numerical differentiation. For IPØINT = 2, a simple first-order difference expression

$$\frac{dM}{dx} = \frac{M-M_p}{\Delta x} \quad (27)$$

is used, where M is the Mach number at the current point and M_p that at the previous point. For IPØINT ≥ 3 , the following second-order difference calculation is used: Let (M,x) be the data for the current point (M_p, x_p) those for the previous point, and (M_{pp}, x_{pp}) those for the point preceding the previous one. A quadratic fit to these data can be written in the form

$$M = c_0 + c_1x + c_2x^2 \quad (28)$$

from which

$$\frac{dM}{dx} = c_1 + 2c_2x \quad (29)$$

Substitution of the three data points into (28) gives, after elimination of c_0 ,

$$\frac{M-M_p}{x-x_p} = c_1 + c_2 (x + x_p) \quad (30a)$$

$$\frac{M-M_{pp}}{x-x_{pp}} = c_1 + c_2 (x + x_{pp}) \quad (30b)$$

Hence

$$c_2 = \frac{\frac{M-M_p}{x-x_p} - \frac{M-M_{pp}}{x-x_{pp}}}{x_p - x_{pp}} \quad (31)$$

Now that c_2 is known, c_1 can be obtained from (30b);

$$c_1 = \frac{M-M_p}{x-x_p} - c_2 (x + x_p) \quad (32)$$

dM/dx is then calculated from (29), and $DLM(L)$ is obtained from

$$\frac{d \ln M}{dx} = \frac{1}{M} \frac{dM}{dx} \quad (33)$$

3.6.5 Calculation of Boundary Layer Thicknesses

Next, the increase in ξ in the step from the previous flow point to the current one is calculated,

$$\Delta \xi = \frac{\Delta x}{\cos b} \quad (34)$$

and ξ is incremented by $\Delta \xi$. The boundary layer integral I (equation I(172)), denoted by $BLINT(L)$, is updated using the trapezoidal integration rule. The derivative $d \ln M/dx$ is converted into $d \ln M/d\xi$ by

$$\frac{d \ln M}{d\xi} = \cos b \frac{d \ln M}{dx} \quad (35)$$

Then n' , equation I(173), is evaluated, and the correlation parameter n is calculated from I(174). The characteristic length L in these equations is taken to be R_0 , defined as the profile radius in the case of a nozzle, and defined by

$$\pi (R_0)^2 = \text{throat area} \quad (36)$$

in the case of a channel. The momentum thickness is calculated from I(184). If the flow geometry is that of an axisymmetric nozzle, the transverse curvature correction I(187) is applied. The smoothed correlation parameter \bar{n} , denoted by XSA(L), is computed from equation I(217).

The effect of free stream Mach number upon the correlations of boundary layer properties is contained in the dependence upon the hypersonic parameter σ , equation I(190). The correlations used in BLAYER are based upon the Dewey-Gross calculations (ref. 5) for a perfect gas. In frozen and nonequilibrium solutions, the parameter σ as calculated from I(190) never approaches unity, even at high Mach numbers, because of the frozen-in dissociation energy in h_0 . To represent the Mach number effect more realistically, the σ defined by I(190) is replaced by an effective value σ_e , defined as

$$\sigma_e = \left[1 + \frac{2}{(\gamma_e - 1) M^2} \right]^{-1} \quad (37)$$

where γ_e is an effective specific heat ratio. For a perfect gas, (37) is equivalent to I(190). In BLAYER, γ_e is calculated as

$$\gamma_e = a^2 W / R_0 T \quad (38)$$

where W is the mixture molecular weight and a the sound speed.

The incompressible form factor HTR(I) is computed from I(189) and the displacement thickness DELBL(L) from I(188). In the case of a nonequilibrium flow solution, the coupling of the boundary layer to the inviscid flow is switched on when the indicator IUPD is set to zero in subroutine THRAT.* The only effect of the coupling in subroutine BLAYER is that the derivatives DDELBL(L) of the displacement thicknesses

*IUPD is always equal to 1 in the frozen and equilibrium solutions.

have to be calculated. To avoid excessive disturbance of the nonequilibrium solution, DDELBL(L) is gradually built up from zero over 29 steps in the flow solution, as explained in Section 7.6 of Volume I.

3.6.6 Heat Flux and Shear Stress

For FINAL = .FALSE., the boundary layer integral BLINT(L) is reset to its value BLINTI at the beginning of the step, and the boundary layer coordinate XI(L) to its initial value XII. Then control is transferred to statement 540, the limit of the DØ loop over L.

For FINAL = .TRUE., the shear parameter XL is calculated from the curvefit I(194), the shear stress TAUW(L) from I(193), the Reynolds analogy factor RAF from I(200), and the heat flux QWDØT(L) from I(195). Then, if L is less than NPRFLS, control is transferred to the end of the DØ loop over L. If L is equal to NPRFLS, several physical quantities are saved for use in the calculations for the next step. When the DØ loop over L has been completed, diagnostic outputs are written if the control variable ISW4B is nonzero, and the RETURN is executed.

3.7 Subroutine BLCALL

During the nonequilibrium solution, all calls to the boundary layer routine BLAYER are made through subroutine BLCALL. Before calling BLAYER, subroutine BLCALL sets up certain data required by BLAYER and by the transport property routines. The species mole fractions are computed from equation I(1) and loaded into the common array SAVEC(J). If the perturbation technique is still being used (INEQ = 0), the perturbations PERTGJ(J) are added to the concentrations $\gamma = GJ(J)$ before equation I(1) is used. In this case, in addition, the temperature perturbation PCT is added to the nondimensional temperature CT and the density perturbation PRHØ is added to RHØ. After BLAYER has been called, the temperature and density are restored to their unperturbed (equilibrium) values. It is not necessary to reset SAVEC(J), because this array is used only in the transport, boundary layer, and model condition calculations, not in the inviscid

flow solution. Indeed, subroutine PRTA (which controls the output of the nonequilibrium flow solution) calls BLCALL to load the SAVEC array even in solutions neglecting the boundary layer (ISW3B = 0), because the output routine ØUT1 obtains the mole fractions from SAVEC.

3.8 Subroutine BXSECT

Subroutine BXSECT is used in searching the array V in common block/TRANS7/for references to the indices of species pairs used in the cross section computations, and also in determining the correspondence between the entries in the V array and the steps in the cross section computations. The argument L of the subroutine is the index of the step in the cross section computations, MV is the index of the last entry in the V array used in step L of the computations, and I and J are the indices of the species pair referenced in V. When the subroutine is called with L = 0, the other arguments MV, I, and J are set internally by the subroutine and need not be furnished by the calling routine; however, all arguments are required when L \neq 0.

In use, subroutine BXSECT is first called with L = 0 to begin the search over the V array. The routine then goes through each step L of the cross section computations, starting with L = 1 and checking for steps with KQ = 9 or 12. The value of MV is accumulated at each step L of the computations from the number of V values NV(KQ(L)) required for that step, where NV(KQ) is a fixed value which has been preset in the code. When a step with KQ = 9 or 12 is found, subroutine BXSECT sets I and J to the indices of the species pair referenced in the V array for that step and returns control to the calling program. The latter program may then reset I and J to any desired values and recall subroutine BXSECT again with the current values of L and MV to enter the revised data in the V array and resume the search of the computations again from the point at which it was left off. It should be noted that for each step with KQ = 12, there are two species pairs referenced in the V array, and returned by subroutine BXSECT, while for KQ = 9 there is only one.

When all steps of the computations have been searched, as indicated by the index L of the step becoming equal to the number of steps NKQ in the edited cross section data in common block/TRANS7/, the subroutine sets L to 0, as a signal that the search has been completed, and returns control to the calling program.

3.9 Subroutine CØMM

The function of subroutine CØMM is to compute some physical quantities upon which the rates of change of the species concentrations and the other dependent variables depend. The actual calculations of the rates of change are then carried out in subroutine EXACT.

CØMM is called only from subroutine DERIVS, and the call to CØMM is always preceded by a call to subroutine THERM, which computes the species properties at the current nondimensional temperature, $CT = T/T_0$. In the case of an electronic nonequilibrium model, DERIVS calls THERM twice to obtain the required properties at both the gas temperature CT and the electron temperature $CTE = T_e/T_0$.

To facilitate the discussion, the correspondence of some of the Fortran symbols with the mathematical notation used in Volume I is reviewed in the following table. Symbols not listed here are defined in Section 4.77.

ALGJ(J)	$\ln \gamma_j$
BETA(I,J)	β_{ij}
CHI(I)	χ_i
CLKF	$\ln k_f$
CLKR	$\ln k_r$
CLNPI(I)	$\ln P_i$
CLNLMC(I)	$\ln (1 - \chi_i)$
CLNTD(L)	$\ln T$ for $L = 1$; $\ln T_e$ for $L = 2$

CLTBF	$\ln (\rho \sum \gamma_j)$, sum over third bodies
CLZ2	$\ln (R_0 T)$ or $\ln (R_0 T_e)$, argument in cm^3 atm/mole
CLZ3	$\ln (\rho R_0 T)$ or $\ln (\rho R_0 T_e)$, argument in atm-g/mole
CTD(L)	T/T_0 for $L = 1$; T_e/T_0 for $L = 2$
EF	ϵ_{fi}
ER	ϵ_{ri}
GJ(J)	γ_j
PI(I)	P_i
PICHI(I)	$P_i \chi_i$
QDPE	\dot{q}_e
QDPR	\dot{q}_r
QF	q_{fi}
QR	q_{ri}
RHØP	ρ
RHPL	$\ln \rho$
SCPGH	$\frac{1}{R_0} \sum_{j=2}^n \gamma_j^{c_{pj}}$
SU	u/u_s
TEP	T_e
T6	$\sum_{i=1}^n \nu_{ij} \ln \gamma_j$ or $\sum_{j=1}^n \nu'_{ij} \ln \gamma_j$

T $\sum \gamma_j$, sum over third-body species

T8 $\sum_{j=1}^n \nu'_{ij}$

T9 $\sum_{j=1}^n \beta_{ij} \left[\ln (R_0 T') + \frac{\mu_j^{\circ}(T')}{R_0 T'} \right]$

where T' is T or T_e

XMJATD(J,L) $\mu_j^{\circ}(T')/R_0 T'$, where T' = T for L = 1,
T' = T_e for L = 2

XNULJ(I,J) ν_{ij}

XNULJP(I,J) ν'_{ij}

Z1 $\ln(1/u)$, for u in cm/sec

Subroutine CØMM may be divided, for purposes of explanation, into three parts: an initialization section, a main loop over the reactions, and a final section. These parts of the subroutine will now be discussed.

3.9.1 Initialization Section

This section extends from the subroutine entry point through the statement RCALC = .FALSE. (card CØM 107). It sets up the ALGJ array, computes a number of sums which are required later in CØMM or in other parts of NATA, calculates the nondimensional flow velocity SU from the energy equation I(245), and calls subroutine GEØM to compute the gas density.

In addition, it contains the controls for intermittent dumping of diagnostic data in the nonequilibrium routines. The indicator for these dumps is the integer variable ISW5B in blank common. ISW5B is preset to zero. For the zero

value, no dumps are produced. If ISW5B is set in the input to a positive value, then dumps are produced each time CØMM, EXACT, RNKT, or PRTA is called. These dumps provide a detailed trace of the operation of the nonequilibrium integration, but give several pages of printed output for each integration step. Thus, it is hardly feasible to run a complete solution with continuous dumps. However, operation of the nonequilibrium routines at late stages in the solution can be studied by calling for intermittent dumps. This is done by setting ISW5B to a negative value in the input. Upon the first entry into CØMM, the variable ICYCLE is then reset from its preset value of 90000000 to |ISW5B|. The counter ICØUNT is incremented by 1 every time CØMM is entered. Whenever ICØUNT is an exact multiple of ICYCLE, ISW5B is reset to 1; otherwise, it is reset to 0. Thus, upon every ICYCLEth entry into CØMM, the dumps are switched on and are produced in all of the nonequilibrium routines until CØMM is entered the next time, when they are switched off again. The coding of these controls is in the first nine executable statements of CØMM.

3.9.2 Main Loop Over the Reactions

The main calculations in CØMM are performed in a large DØ loop headed by the statement, DØ 460 I = 1, ISR (card CØM 109). Figure 7 is a flowchart of this main loop. In essence, this loop performs only two functions:

- 1) First, it computes the quantities $P_i \chi_i$ appearing in the rate equations I(287).
- 2) Second, in the case of an electronic nonequilibrium model, it evaluates the energy transfer to the electron gas \dot{q}_e and the radiative loss \dot{q}_r due to the reactions.

The quantities P_i and χ_i are computed from equations I(288) and I(289), respectively, in logarithmic form:

$$\ln P_i = \ln\left(\frac{1}{u}\right) + \ln k_{fi} + (\nu_i - 1) \ln \rho + \sum_{j=1}^n \nu_{ij} \ln \gamma_j \quad (39)$$

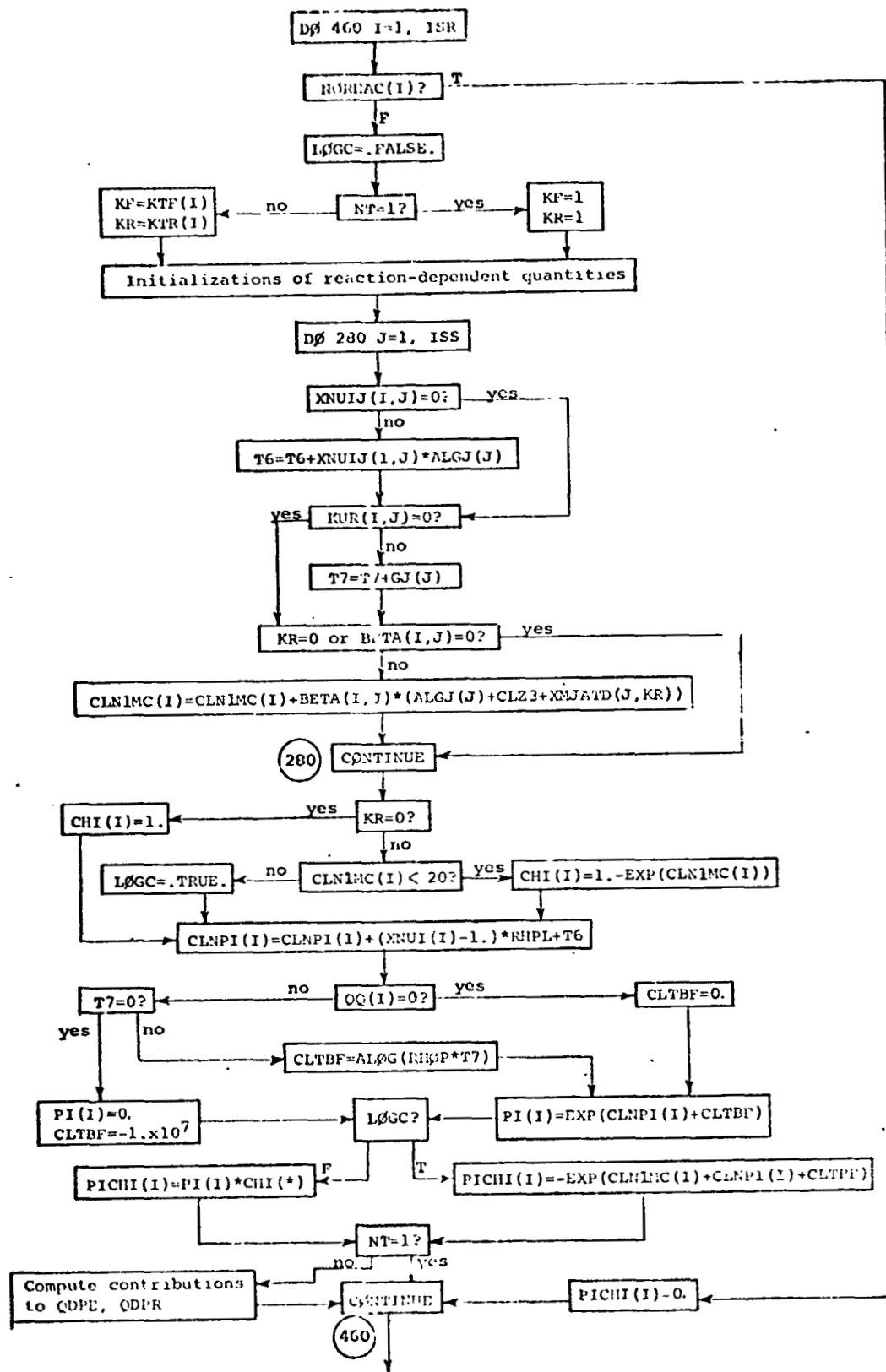


Figure 7. Flow Chart of Main Loop Over Reactions in Subroutine CQMM

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$$\ln (1 - \chi_i) = \beta_i \ln (\rho R_0 T) + \sum_{j=1}^n \beta_{ij} \left[\ln \gamma_j + \frac{\mu_j^0}{R_0 T} \right] \quad (40)$$

In equation (40), the equilibrium constant K_i in I(289) has been expressed in terms of more basic quantities using equation I(278). In the summations indicated in (39) and (40), terms in which the stoichiometric factor ν_{ij} or β_{ij} is zero are omitted.

If $\ln (1 - \chi_i)$ is less than 20, P_i and χ_i are computed from (39) and (40) using

$$P_i = \exp (\ln P_i) \quad (41)$$

$$\chi_i = 1 - \exp [\ln (1 - \chi_i)] \quad (42)$$

and $P_i \chi_i$ is obtained by multiplication. On the other hand, if $\ln (1 - \chi_i)$ is greater than or equal to 20, so that $(1 - \chi_i)$ is greater than 4.85×10^8 , the term 1 in $(1 - \chi_i)$ is neglected and the product $P_i \chi_i$ is computed as

$$P_i \chi_i = - \exp [\ln P_i + \ln (1 - \chi_i)] \quad (43)$$

This algorithm allows the computation of a finite $P_i \chi_i$ value in situations where P_i is so small that an exponential underflow occurs in (41), but χ_i is negative and so large that $P_i \chi_i$ is within the range representable by floating-point numbers. It also makes possible a simple treatment of the special cases which arise when one or more of the γ_j are zero, as discussed below.

If a list of third-body (catalytic) species has been specified for the reaction, the expression I(288) for P_i is replaced by

$$P_i = \left[\frac{\rho^{\nu_i - 1}}{u} k_{fi} \prod_{j=1}^n \gamma_j^{\nu_{ij}} \right] \cdot \rho \sum_{tb} \gamma_k \quad (44)$$

in which the sum on the right is over the third-body species specified. The corresponding logarithmic expression is (39) with a term $\ln(\rho \sum_{\text{tb}} \gamma_k)$ added to the right. The third bodies do not affect χ_i . Equation (44) may be derived from I(283) and I(287) by noting that, for a catalytic species (k), $\nu_{ik} = \nu'_{ik} = 1$. Thus, a factor $\rho \gamma_k$ can be removed from both terms on the right in I(283). If the i^{th} reaction involves more than one third-body species, each such species gives a separate three-body reaction. The effects of these reactions can be summed to give a single reaction with an effective third-body concentration which is the sum of the concentrations of the actual catalytic species, as in (44).

In the case of an electronic nonequilibrium gas model, there are two temperatures, the gas temperature T and the electron temperature T_e . The forward reaction rate k_{fi} can depend on either T or T_e , or both, or neither. The form of temperature dependence of k_{fi} is specified by the array $\text{KTF}(\text{I})$. If the indicator $\text{KF} = \text{KTF}(\text{I})$ is equal to 1, k_{fi} is given by I(69). If KF is equal to 2, the temperature T in I(69) is replaced by T_e . This procedure is coded by placing the two non-dimensional temperatures into a two-element array $\text{CTD}(\text{L})$, and placing the logarithms of the two temperatures (expressed in degrees Kelvin) into an array $\text{CLNTD}(\text{L})$. Both cases are then treated by a single formula involving $\text{CTD}(\text{KF})$ and $\text{CLNTD}(\text{KF})$, i.e., statement 190 of CØMM , which evaluates $\ln k_{fi}$ from

$$\ln k_{fi} = \ln C_i + \eta_i \ln T_k - \frac{E_{ai}}{R_0 T_k} \quad (45)$$

where T_k is equal to T for $k = 1$ and equal to T_e for $k = 2$. In (45), $\ln C_i = \ln A_i - \eta_i \ln(10^4)$.

Three special forms for the forward reaction rate, used in the electronic nonequilibrium model for argon (Appendix A in Volume II), are indicated by $\text{KTF}(\text{I}) = 3, 4, \text{ and } 5$. For $\text{KTF}(\text{I}) = 3$, k_f is assumed to be given by

$$k_f = C_i T_e^{\eta_i} (1 - e^{-E_i/R_0 T}) \quad (46)$$

where E_i is a parameter with dimensions of energy. For $KTF(I) = 4$,

$$k_f = \frac{C_i T_e^{\eta_i}}{\max(1, \tau)} \quad (47)$$

where

$$\tau = b \rho \gamma_a R \quad (48)$$

Here R is the local nozzle radius (or equivalent radius in the case of a channel), and γ_a is the concentration of the product atom in the reaction. For $KTF(I) = 5$,

$$k_f = \frac{C_i}{\sqrt{R}} \quad (49)$$

In NATA, the reverse rate constant k_{ri} is calculated from the detailed-balancing relation I(277), using the equilibrium constant K_i , equation I(278). In electronic non-equilibrium gas models, the coding of CØMM allows for three types of temperature dependence of K_i . If the indicator $KR = KTF(I)$ is zero, the reverse reaction is assumed to occur only at a negligible rate. If KR is 1, K_i is calculated from I(278) using the gas temperature T . If $IR = 2$, the temperature T in I(278) is replaced by T_e . The chemical potentials μ_j° in I(278) are temperature dependent. During calculations based on an electronic nonequilibrium model, both $\mu_j^{\circ}(T)$ and $\mu_j^{\circ}(T_e)$ are computed, and these data are stored in the doubly dimensioned array $XMJAT(J,L)$ with dimensions (20,2). The coding of equation (40) utilizes the array with the index $L = KR$ so that both cases can be treated using a single formula, card CØM 153.

Cases in which one or more of the concentrations $\gamma_j = 0$ require some special consideration because of the appearance of $\ln \gamma_j$ in the formulas used in CØMM. In NATA, the γ_j are represented by the double precision array $GJ(J)$. A zero

value for an element of this array could come about in one of two ways:

- (1) One of the concentrations $GJ(J)$ could underflow as a result of its decrease during the flow solution. However, this is most unlikely to occur in practice because, to underflow, $GJ(J)$ must fall below 10^{-78} (on the IBM 360/75) or below 10^{-308} (on the UNIVAC 1108). Since $GJ(J)$ can decrease by no more than 10 percent in a single integration step because of step size controls, the number of steps required to reach underflow from initial reservoir values of (say) 10^{-1} to 10^{-20} is of the order of 1000 or more, even on the IBM 360. Also, unless the control parameter $GAMIN$ were set to zero, the decrease of $GJ(J)$ would be stopped, long before underflow was reached, by the freezing of minor species (Section 7.5.3 of Volume I).
- (2) A zero $GJ(J)$ value could arise accidentally, at an intermediate point in the Treanor-Runge-Kutta integration step, as a result of a negative $\Delta \gamma_j$ which happened to be exactly equal in magnitude to the γ_j value at the start of the step. This is quite unlikely, but it could happen.

To allow for these rather remote possibilities, the coding of $CØMM$ is designed to cope with zero concentrations if they should occur. If one of the species concentrations, γ_k , is zero, four special cases can arise, depending upon the values of the stoichiometric coefficients ν_{ik} , ν'_{ik} in the reaction I(219):

$$\text{Case 1: } \nu_{ik} = \nu'_{ik} = 0 \quad (50a)$$

$$\text{Case 2: } \nu_{ik} = 0, \quad \nu'_{ik} > 0 \quad (50b)$$

$$\text{Case 3: } \nu_{ik} > 0, \nu'_{ik} = 0 \quad (50c)$$

$$\text{Case 4: } \nu_{ik} > 0, \nu'_{ik} > 0 \quad (50d)$$

In the Cornell Aeronautical Laboratory program from which NATA was derived, each of these cases was treated separately by special coding. In the present version of the code, all of these cases are treated using the formulas (39) to (44) with the aid of a computational artifice.

Cases involving zero species concentrations γ_j require special consideration because of the appearance of $\ln \gamma_j$ in the formulas (39) and (40). Any attempt of the program to evaluate $\ln 0$ would lead to immediate termination of the run by the computer operating system. In subroutine CØMM, the quantities $\ln \gamma_j$ are stored in an array, ALGJ(J). For species whose concentrations γ_j are zero, ALGJ(J) is set to a large negative number, L, whose magnitude is chosen in accordance with the following criteria:

- (1) L must be sufficiently large in magnitude that, even when combined additively with other terms as indicated in (39) and (40), it is sufficiently negative to produce a guaranteed exponential underflow, i.e.,

$$\exp(L + \text{other terms}) = 0 \quad (\text{exactly}) \quad (51a)$$

This will happen so long as the argument of the exponential function is less than about -88 on the UNIVAC 1108 or less than about -180 on the IBM 360/75.

- (2) L must be sufficiently small in magnitude that the operation

$$L + \text{other terms} - L = \text{other terms} \quad (51b)$$

gives a result for "other terms" containing several significant digits of accuracy. The "other

terms" in equations (39) and (40) are all natural logarithms of the values of physical quantities whose values lie roughly in the range 10^{-20} to 10^{20} ; thus, the logarithms range from about -50 to +50. The second condition can thus be satisfied by taking L to be less than about 10^4 or 10^5 . If L were chosen to be much larger, say 10^{10} , then the first addition in (1b) would give L + other term = L (exactly) on the UNIVAC 1108, because the change in L due to addition of the "other terms" would be less than the accuracy with which L can be represented as a floating point number (about 8 significant figures).

In subroutine COMM, L has been chosen to be -1.00, a value which clearly satisfies both conditions.

Now assume that a particular concentration γ_k is zero. With $\ln \gamma_k = L$, the formulas (39) and (40) treat the four cases (50) as follows:

In Case 1, ν_{ik} and β_{ik} are both zero, so that the term $\ln \gamma_k$ does not actually appear in either (39) or (40). In this case, according to (50a), the species k whose concentration γ_k is zero does not participate in the reaction as either a reactant or a product.

In Case 2, $\ln \gamma_k$ does not appear in (39), so that the evaluation of $\ln P_i$ proceeds normally. However, $\beta_{ik} = \nu'_{ik}$ is positive, so that the right hand side of (40) contains a term $\nu'_{ik} L$ which is negative and large in magnitude. The resulting exponential underflow during the evaluation of (42) then gives $\chi_i = 1$. This is the correct result for this case, since $\gamma_k^{\beta_{ik}}$ is a zero factor in the second term on the right in equation I(289).

In Case 3, equations (39) and (40) can be written

$$\ln P_i = \ln \left(\frac{1}{U} \right) + \ln k_{fi} + (\nu_i - 1) \ln \rho + \sum_{j \neq k} \nu_{ij} \ln \gamma_j + \nu_{ik} L \quad (52a)$$

$$\ln(1 - \chi_i) = \beta_i \ln(\rho R_0 T) + \sum_{j \neq k} \beta_{ij} \left[\ln \gamma_j + \frac{\mu_j^0}{R_0 T} \right] - \nu_{ik} \left(L + \frac{\mu_k^0}{R_0 T} \right) \quad (52b)$$

Here the term $-\nu_{ik} L$ on the right in (52b) is positive and very large; thus, equation (42) is not used and the product $P_i \chi_i$ is evaluated using (43), which gives

$$P_i \chi_i = - \exp \left\{ \ln \left(\frac{1}{u} \right) + \ln k_{fi} + (\nu_i - 1) \ln \rho + \beta_i \ln(\rho R_0 T) + \sum_{j \neq k} \nu'_{ij} \ln \gamma_j + \sum_{j=1}^n \beta_{ij} \frac{\mu_j^0}{R_0 T} \right\} \quad (53)$$

$$= - \frac{k_{fi} \rho^{\beta_i + \nu_i - 1}}{K_i} \prod_{j \neq k} \gamma_j^{\nu'_{ij}}$$

The terms involving L have cancelled out. The validity of equation (53) for Case 3 can be verified directly by multiplying equation I(288) and I(289), letting $\gamma_k \rightarrow 0$, and taking (50c) into account.

In Case 4, equations (39) and (40) give

$$\ln P_i = \ln \left(\frac{1}{u} \right) + \ln k_{fi} + (\nu_i - 1) \ln \rho + \sum_{j \neq k} \nu_{ij} \ln \gamma_j + \nu_{ik} L \quad (54a)$$

$$\ln(1 - \chi_i) = \beta_i \ln(\rho R_0 T) + \sum_{j \neq k} \beta_{ij} \left[\ln \gamma_j + \frac{\mu_j^0}{R_0 T} \right] + (\nu'_{ik} - \nu_{ik}) \left[L + \frac{\mu_k^0}{R_0 T} \right] \quad (54b)$$

Addition of these two equations gives

$$\begin{aligned} \ln [P_i (1 - \chi_i)] &= \ln \left(\frac{1}{u}\right) + \ln k_{fi} + (\nu_i - 1) \ln \rho + \beta_i \ln (\rho R_0 T) \\ &+ \sum_{j \neq k} \nu'_{ij} \ln \gamma_j + \sum_{j \neq k} \beta_{ij} \frac{\mu_j^0}{R_0 T} + \beta_{ik} \frac{\mu_k^0}{R_0 T} \\ &+ \nu'_{ik} L \end{aligned} \quad (55)$$

Since $\nu'_{ik} L$ is negative and large in magnitude, (55) implies that

$$P_i (1 - \chi_i) = P_i - P_i \chi_i = 0 \quad (56)$$

Also, (54a) and (41) give $P_i = 0$; thus, $P_i \chi_i = 0$. This result is correct for Case 4, as may be seen by setting one of the γ_k to zero in I(286) and taking (50d) into account.

The preceding analysis can easily be generalized to include cases in which more than one of the concentrations is zero. Thus, the artifice of setting $\ln \gamma_k = L$ when $\gamma_k = 0$ provides a satisfactory treatment of all cases that can arise.

In calculations for a chemical nonequilibrium model ($NT = 1$), the main loop provides only calculations of P_i and $P_i \chi_i$. In calculations for an electronic nonequilibrium model the loop also computes the contributions to the energy transfer to the electron gas (QDPE) and the radiative power loss (QDPR). These computations are provided by the coding below statement 390, and simply consist of evaluation of the formulas

$$(\Delta \dot{q}_e)_i = k_{fi} \epsilon_{fi} \prod_{j=1}^n (\rho \gamma_j)^{\nu'_{ij}} - k_{ri} \epsilon_{ri} \prod_{j=1}^n (\rho \gamma_j)^{\nu'_{ij}} \quad (57a)$$

$$(\Delta \dot{q}_r)_i = k_{fi} q_{fi} \prod_{j=1}^n (\rho \gamma_j)^{\nu_{ij}} - k_{ri} q_{ri} \prod_{j=1}^n (\rho \gamma_j)^{\nu'_{ij}} \quad (57b)$$

which are obtained by combining equations I(320) and I(32lb,c). In equations (57), $(\Delta \dot{q}_e)_i$ and $(\Delta \dot{q}_r)_i$ denote the contributions of the i th reaction to \dot{q}_e and \dot{q}_r , respectively. The energy partition parameters ϵ_{fi} , ϵ_{ri} , q_{fi} , q_{ri} are obtained by calling subroutine EPART. Effects of third-body species, if any, are taken into account using the quantity $CLTBF = \ln(\rho \sum_{tb} \gamma_j)$, already evaluated during the computation of $P_i \chi_i$.

3.9.3 Final Section

After the main loop over the reactions has been completed, CØMM checks the value of INEQ. If this indicator is zero (i.e., if the perturbation technique is still being used to generate the flow solution), the RETURN is executed immediately. If INEQ is nonzero (i.e., if the numerical integration of the rate equations has been started), CØMM calculates the non-dimensional pressure, $PRES = p/p_0$. If a chemical nonequilibrium model is being used, the pressure is computed from equation I(273) in the form

$$\frac{p}{p_0} = \frac{\rho}{\rho_0} \frac{T}{T_0} \frac{W_0}{W} \quad (58)$$

If an electronic nonequilibrium model is in use, equation I(323) is used to calculate the pressure in the form

$$\frac{p}{p_0} = \frac{\rho}{\rho_0} \cdot \frac{\gamma_e T_e + \gamma_h T}{T_0} \cdot W_0 \quad (59)$$

In either case, p/p_0 is then divided by $RØBARA = \tilde{\rho}_0 / \rho_0$ to eliminate the effects of gas imperfections, if any, upon the value of the reservoir density, ρ_0 . This is done because gas imperfections are not taken into account in the nonequilibrium integration, as explained in Section 2.5 of Volume I. In addition, the gas entropy is calculated from equation I(251).

If an electronic nonequilibrium model is in use, CØMM computes the contribution to the electron energy transfer \dot{q}_e from elastic collisions, using the formula

$$(\Delta \dot{q}_e)_{\text{elas}} = \sum_{j=2}^n \frac{2W_e}{W_j} \cdot \frac{3}{2} R_0(T - T_e) \frac{n_e \nu_{ej}}{N_0} \quad (60)$$

where for neutral species j , the collision frequency ν_{ej} is given by

$$\nu_{ej} = \frac{4}{3} \sqrt{\frac{8 R_0 T_e}{\pi W_e}} n_j \bar{Q}_{ej}^{-(1,1)} \quad (61)$$

and for ionic species

$$\nu_{ei} = \frac{8}{3} \sqrt{\frac{\pi}{m_e}} \frac{n_i e^4}{(2k T_e)^{3/2}} \ln \left(\frac{k^3 T_e^2}{\pi n_e e^6} \right) \quad (62)$$

These formulas are derived in Appendix A of Volume II; see equations II(25-31). The notation is

W_e, W_j	molecular weights for electron and j^{th} species
$n_e = N_0 \rho \gamma_e$	electron concentration
n_j	concentration of a neutral atom or molecule
n_i	concentration of an ion
$\bar{Q}_{ej}^{(1,1)}$	Maxwell-averaged momentum transfer cross section (assumed the same for all neutrals)
m_e	electron mass
k	Boltzmann constant
N_0	Avogadro number
e	electronic charge (esu)

In NATA, the cross section $\bar{Q}_{ej}^{(1,1)}$ is obtained by calling the function PIØMEG(TEP). The constant factors in (61) and (62) are represented by VIC1, VIC2, and VNC, and are preset in a data statement in CØMM. Equation (60) is evaluated straightforwardly, and the result QELAS = $(\Delta \dot{q}_e)_{elas}$ is added to the \dot{q}_e due to the reactions to obtain the total rate of energy transfer to the electron gas.

3.10 Subroutine CXSECT

Subroutine CXSECT is used by subroutine XSECT to determine the correspondence between species pairs in the master species list for the code and in the rearranged species list stored in the common variable I(J) (see the discussion of subroutine XSECT below). When the subroutine is called with its arguments set equal to the indices of a pair of species in the master species list, it returns the arguments as the indices of the corresponding species pair in the rearranged list. If the original indices are denoted by L,J, the corresponding revised indices are I(L), I(J) if L-J and I(L)-I(J) have the same sign, and are I(J), I(L) otherwise.

3.11 Subroutine DERIVS

During the nonequilibrium solution, the derivatives of the dependent variables, including the species concentrations, are calculated by calling subroutines THERM, CØMM, and EXACT in sequence. If an electronic nonequilibrium gas model is in use, THERM must be called twice, once with CT = T/T₀, and once with CT = T_e/T₀, to provide species properties at both the gas temperature and the electron temperature. The use of these properties has been explained in the analysis of subroutine CØMM. Also, if the boundary layer is being included in the flow solution, an iteration is necessary to determine the self-consistent solution for the boundary layer displacement thickness δ^* and the derivatives of the flow variables (Section 7.6 of Volume I). These calculations are called for at several locations in subroutines NØNEQ and RNKT. For compactness of coding, these operations are all placed in a separate subroutine, DERIVS.

For an electronic nonequilibrium model (TN = 2), DERIVS saves the current value of CT in the location CTSAVE, resets CT to $CTE = T_e/T_0$, and calls THERM to compute the species properties at the electron temperature. The values $\mu_i^0(T_e)/R_0T_e$ are stored directly into XMJATD(I,2). The enthalpy, specific heat, and entropy of the electrons are saved temporarily in the locations HJE, CPJE, and SENTE. Then CT is reset to the value T/T_0 and THERM is called again to compute the species properties at the gas temperature. THERM overwrites the previously calculated data for the electrons (species 1), but these are reset from HJE, CPJE, and SENTE.

For a chemical nonequilibrium model (NT = 1), THERM is called only once, to obtain the properties for all the species at the gas temperature, T.

In either case, DERIVS next calls CØMM and EXACT to compute the derivatives of the species concentrations and other dependent variables. It then applies the condition I(407b); if this condition is violated, the indicator FAILED is set to .TRUE. and a RETURN is executed.

The remaining portion of DERIVS controls the iteration to obtain a self-consistent solution for the boundary layer displacement thickness and the derivative $d \ln M/dx$ of the logarithm of the Mach number. If the boundary layer is not being included in the flow solution (ISW3B = 0), or if the solution is still in the upstream (uncoupled) region (IUPD = 1), this iteration is not performed. The technique used in the iteration is explained in Section 7.6 of Volume I, in the vicinity of equation I(424).

3.12 Subroutine DSMSØL

DSMSØL is the subroutine called for simultaneous solution of systems of linear equations in the calculation of thermochemical equilibrium conditions, in the equilibrium flow solution, and in the nonequilibrium solution. The calling sequence for DSMSØL was embodied in the original program version received from Cornell Aeronautical Laboratory (reference 6), but the subroutine itself was not provided by CAL because it was written in machine language and would not be operable on other types of computer system. To provide the required

capability for solution of systems of equations, subroutine SIMQ from the IBM Scientific Subroutine Package (reference 7) was incorporated into NATA, and a Fortran DSMSØL routine was written to rearrange the data for the equations from the form provided by the CAL routines into the form required by SIMQ.

For a system of linear equations

$$\sum_{k=1}^i a_{jk} X_k = b_j \quad (j = 1, \dots, i) \quad (63)$$

in i unknowns ($X_1 \dots, X_i$), NATA stores the coefficients a_{jk} and constants b_j in the double precision array AA in unlabelled common. The dimensions of AA are (22,24). The matrix of coefficients a_{jk} is stored as

$$a_{jk} = AA(j,k) \quad \begin{cases} j = 1, \dots, i \\ k = 1, \dots, i \end{cases} \quad (64)$$

The constants b_j are stored as

$$b_j = AA(j, i + 1) \quad (65)$$

Subroutine SIMQ (discussed below) requires that the coefficients a_{jk} be stored columnwise in a singly dimensioned array A, without gaps, and that the constants b_j be stored in an array B. DSMSØL reorganizes the data into this form, using the storage locations of the array AA to contain A and B. The array A is assigned to the first i^2 storage locations in AA, that is, AA(1,1), AA(2,1), ..., A(22,1), AA(1,2), AA(2,2), ... The array B is assigned the first i storage locations in the 23rd column of AA, that is, AA(1,23), ..., AA(i,23). If the number of equations i is equal to 22, the reorganization leaves the matrix AA unaltered. If i is less than 22, the coefficients a_{jk} are shifted down so as to fill up the first columns of AA without gaps, and the constants b_j are shifted

up into the previously unused 23rd column. After SIMQ has been called, the solution values x_k provided by SIMQ are shifted from the 23rd column into the $(i + 1)$ th column, for use by the calling routine.

The coding of DSMSØL is straightforward, and can be followed easily with the help of the definitions given in the glossary of Fortran symbols (Section 4.80).

3.13 Subroutine DUMP

DUMP is part of the NATA system for dealing with possible errors in code operation. NATA contains numerous validity checks. If one of these is violated, the routine in which the error was detected calls DUMP. This small subroutine sets the indicator ERR to .TRUE. and writes a message giving the name RNAME of the calling routine. It then returns control to the calling routine. The calling routine then executes a RETURN. At each higher-level routine, statements of the form IF(ERR) RETURN pass control up is still higher levels until the main program is reached. In MAIN, when a .TRUE. value of ERR is detected, subroutine DUMPEX is called to print diagnostic data (see below). After the return from DUMPEX, MAIN proceeds as if a normal case completion had occurred; i.e., the job is not terminated unless the input ISW4A is zero.

Originally, the operations now performed by DUMP and DUMPEX were included in a single subroutine (called DUMP). When NATA become so large that it had to be run using overlay on the UNIVAC 1108, the original DUMP routine was divided into two parts, and the larger of these (DUMPEX) was placed into a separate overlay segment so as to minimize the core storage requirement.

3.14 Subroutine DUMPEX

DUMPEX produces the diagnostic dumps when an error has been detected in the operation of the code. The routine contains unlabelled common and many of the labelled common blocks used in NATA. When DUMPEX is executed, most of the variables and arrays listed in these common statements are printed out

in namelist format. In addition, if a binary tape of data for plotting is being written (DATAPE = .TRUE.), DUMPEX backspaces this tape to eliminate all of the records written during the current case.

3.15 Subroutine ELCØND

Subroutine ELCØND computes the electrical conductivity SIGMA of the gas in mhos/cm from equation I(98) using the values of $SSIG = (e/k)^2 / \sqrt{T}$ and $Q(1, I, J) = k^{-1} \sqrt{T} \Delta_{ij}^{(1)}$ computed in subroutine TRANSP. In this computation it is assumed that if electrons are present in the gas, they are the first species. If, on the other hand, electrons are not present (IELEC = 0), the electrical conductivity is set to 0.0 and the computation is bypassed.

3.16 Subroutine ELTIME

The function of subroutine ELTIME is to determine and print out the elapsed execution time since the beginning of the run. When NATA is run on the UNIVAC 1108, ELTIME uses the utility routines .SET and TIME, which are provided by the EXEC II operating system. When the code is used on the IBM 360/75 at Avco, the functions of RESET and TIME are simulated by a small Fortran program (called RESET, Sec. 3.58) which utilizes the system subroutine ACUCPU.

ELTIME contains a logical indicator CALLED, which is preset to .FALSE. in a data statement. When ELTIME is called for the first time in a job, a test on CALLED leads to a call to RESET, which initializes the time counter I for subroutine TIME to 0, and ETØ and ET are both set to zero. Also, CALLED is set to .TRUE., so that this part of the routine is bypassed when ELTIME is called on subsequent occasions in the run.

In such subsequent calls, ELTIME sets ETP = ETØ, and then sets ETØ to ET, where ET is the value left in storage by the previous call (unless this argument has been altered by the calling routine). Then ELTIME calls subroutine TIME(I), which returns the elapsed time I since the call to RESET, measured in milliseconds. ELTIME converts this time to seconds

and stores it in ET. Then, unless the argument IP is zero, ELTIME converts ET into minutes and stores the result in ETM, and also computes the elapsed time DET in seconds since the last previous printed output of the time. A message is then printed giving ETM and DET. If the printed message is skipped because IP = 0, ETØ is reset to ETP to maintain its significance as the elapsed time since the printing of the last time message.

3.17 Subroutine EPART

This subroutine computes the parameters governing the partition of reaction energy between radiative loss and energy transfer to the electron gas for each reaction in a gas model which includes electronic nonequilibrium. The arguments in the calling sequence are defined in Section 4.85. The subroutine treats six cases, or types of reactions, which are distinguished by an index IT. In the following discussion of these cases, $\epsilon_f = EF$ and $-\epsilon_r = -ER$ denote the energies gained by the electron gas in one mole of reactions in the forward and reverse directions, respectively; and $q_f = QF$ and $-q_r = -QR$ denote the corresponding energies lost by radiation.

The six cases are

$$IT = 1 \quad \epsilon_f = -a R_0 T_e$$

$$q_f = \epsilon_0 - \epsilon_f$$

$$\epsilon_r = q_r = 0$$

$$IT = 2 \quad \epsilon_f = -\frac{3}{2} R_0 T_e$$

$$q_f = \epsilon_0 - \epsilon_f$$

$$\epsilon_r = q_r = 0$$

$$IT = 3 \quad \epsilon_f = \epsilon_r = q_f = q_r = 0$$

$$IT = 4 \quad \epsilon_f = \epsilon_r = -\frac{3}{2} R_0 T_e$$

$$q_f = q_r = 0$$

$$IT = 5 \quad \epsilon_f = \epsilon_r = \epsilon_0$$

$$q_f = q_r = 0$$

$$IT = 6 \quad q_f = \epsilon_0$$

$$\epsilon_f = \epsilon_r = q_r = 0$$

In these formulas, $\epsilon_0 = E_0$ denotes the energy available for partitioning between⁰the electrons and the radiative losses, R_0 the universal gas constant (1.9872 cal/mole-deg), and T_e the electron temperature ($^{\circ}$ K).

3.18 Subroutine EQCALC

EQCALC computes the conditions of thermochemical equilibrium at specified temperature T ($^{\circ}$ K) and pressure P (atm), using the method explained in Section 6.1 of Volume I. As an aid to following the coding, the correspondence of Fortran and mathematical symbols is reviewed in the following list:

BET(I)	$\nu_{i-c}^* - 1$	(I = i-c)
CDIJ(I,J)	$\bar{\nu}_{i-c,j}$	(I = i-c)
CGI(I)	W_i	
QM(I)	q_i	

XMJAT(I) μ_i°/R_0T
 ZCAP(I) X_i
 ZGMU(I) $\ln X_i$
 ZPZ $\ln p$

EQCALC first saves the current value of the nondimensional temperature CT in the location CTSAVE. This is done because EQCALC may be called with temperatures T differing from the temperature at the current flow point (for example, during the equilibrium normal shock solution). The nondimensional temperature CT is then reset to T/T₀ to allow calculation of the species thermal properties by calling subroutine THERM.

Next, EQCALC computes initial estimates of the species mole fractions and related quantities. The mole fractions for the independent species (I = 1 to ISC) are set to $q_i = QM(I)$, defined by equation I(10). If any of the QM(I) are zero or negative (due to roundoff errors), the corresponding ZCAP(I) values are reset to 10⁻³. Subroutine THERM is then called to compute the thermal properties of all the species at temperature T. The mole fractions for the dependent species (I = ISC + 1 to ISS) are then estimated using equation I(224), which can be written in the logarithmic form

$$\ln X_i = CHII(i-c) + \sum_{j=1}^c \bar{v}_{i-c,j} \ln X_j \quad (66)$$

where

$$CHII(i-c) = -\frac{\mu_i^{\circ}}{R_0T} + (\nu_{i-c}^* - 1) \ln p + \sum_{j=1}^c \bar{v}_{i-c,j} \frac{\mu_j^{\circ}}{R_0T} \quad (67)$$

In this calculation, SKIL(i) is used as an intermediate variable, first to represent $\ln X_{i-c}$, then X_{i-c} . If the number of dependent species, ISMC = n-c, is zero, then the calculation of

the dependent species mole fractions is skipped here and elsewhere in EQCALC.

Once initial estimates of the X_i and $\ln X_i$ have thus been determined, EQCALC enters the Newton-Raphson iteration based on equations I(277) to I(233). The iterative loop runs from statement 70 down to (but not including) statement 220. The constant terms F_j , equation I(227), in the system of linear equations I(233a) for the corrections h_n^r are loaded into the $M1^{\text{th}}$ column of the matrix AA by the $D\emptyset$ loop ending at statement 90. Note that, in statement 80, X_i for $i = c + 1$ to n is denoted by SKIL(i). Then the matrix of coefficients $-X_n^r (\partial F_j / \partial X_n)^r$, equation I(233b), is loaded into the first ISC rows and columns of AA by the nest of $D\emptyset$ loops ending at statement 140. Subroutine DSMSØL is then called to solve the system of equations for the correction factors h_n^r . Upon the return from DSMSØL, the solution values for these quantities are obtained from the $M1^{\text{th}}$ column of the matrix AA, where $M1 = \text{ISC} + 1$.

The corrected values of the species mole fractions are then calculated. The mole fractions for the independent species ($i = 1$ to c) are obtained using equation I(234):

$$X_k^{r+1} = (1 + h_k^r) \cdot X_k^r \quad (68)$$

The factor $1 + h_k^r$ is denoted by ZA. If ZA is zero or negative, the corrected mole fraction is estimated by halving the previous value:

$$X_k^{r+1} = X_k^r / 2 \quad (69)$$

These calculations are done in the $D\emptyset$ loop ending at statement 180.

The corrected mole fractions for the dependent species are calculated from the equilibrium relations I(224), based on the corrected X_i for the independent species. Equations (66) and (67) are again used, in the $D\emptyset$ loop ending at statement 200.

The convergence test for the Newton-Raphson iteration is that the absolute values of all the h_k^r be less than or equal to a criterion value TEST, which is preset (in subroutine INIT) to 10^{-6} . If convergence has not been achieved after NTEST = 100 iterations, the DUM routine is called to terminate the case and produce diagnostic output.

When convergence has been achieved, EQCALC computes a number of gas properties based on the temperature T, the pressure P, and the equilibrium mole fractions determined by the Newton-Raphson solution of equations I(227). The nondimensional molar entropy of the gas mixture, $ZSEN = S/R_0$, is calculated using equation I(30b). Note that the specific entropy of the gas is equal to

$$s = \left(\frac{S}{R_0}\right) \cdot \frac{R_0}{W} = \frac{ZSEN \cdot CRA}{ZCM} \quad (70)$$

in cal/gm °K. The nondimensional molar enthalpy ZCH of the gas is calculated from equation I(238) and the molecular weight ZCM from equation I(236). Finally, the corrections for gas imperfections, equations I(79) and I(81), are applied.

3.19 Subroutine EXACT

This subroutine computes the derivatives of the species concentrations and the other dependent flow variables for use in the nonequilibrium solution. EXACT consists of two sections. The first section (down to statement 100) calculates the derivatives required in the perturbation solution; the second gives those used in the nonequilibrium integration.

3.19.1 Derivatives for the Perturbation Solution

The derivatives required in the perturbation solution (Section 7.3 of Vol. I) are those of the equilibrium flow variables. These derivatives are computed in EXACT by solving a system of $n+2$ linear equations of the form

$$\sum_{j=1}^{n+2} A_{ij} X_j = B_i \quad (71)$$

The unknowns X_j are

$$X_j = \frac{d \bar{\gamma}_j}{dx} \quad \text{for } j = 1 \text{ to } n \quad (72a)$$

$$X_{n+1} = \frac{1}{T_0} \frac{dT}{dx} \quad (72b)$$

$$X_{n+2} = \frac{d \ln \bar{\rho}}{dx} \quad (72c)$$

The system of equations is

$$i=1 \text{ to } c: \quad \sum_{j=1}^n \alpha_{ji} \frac{d \bar{\gamma}_j}{dx} = 0 \quad (73a)$$

$$i=c+1 \text{ to } n: \quad \sum_{j=1}^c \frac{\bar{v}_{i-c,j}}{\bar{\gamma}_i} \frac{d \bar{\gamma}_j}{dx} - \frac{1}{\bar{\gamma}_i} \frac{d \bar{\gamma}_i}{dx} + \frac{T_0}{\bar{T}} \left[\frac{\bar{H}_i/R_0 T_0}{\bar{T}/T_0} - 1 - \sum_{j=1}^c \bar{v}_{i-c,j} \left(\frac{\bar{H}_j/R_0 T_0}{\bar{T}/T_0} - 1 \right) \right] \frac{d(\bar{T}/T_0)}{dx} + (\gamma_{i-c}^* - 1) \cdot \frac{d \ln \bar{\rho}}{dx} = 0 \quad (73b)$$

$$i=n+1: \quad \sum_{j=1}^n \frac{d \bar{\gamma}_j}{dx} + \frac{T_0}{\bar{T}} \left[\sum_{j=1}^n \bar{\gamma}_j \right] \frac{d(\bar{T}/T_0)}{dx} + \left[\frac{1}{W} - \frac{u^2}{u_s^2} \cdot \frac{u_s^2}{R_0 T_0} \cdot \frac{T_0}{\bar{T}} \right] \frac{d \ln \bar{\rho}}{dx} = \frac{\pi^2}{u_s^2} \cdot \frac{u_s^2}{R_0 T_0} \cdot \frac{T_0}{\bar{T}} \cdot \frac{d \ln A}{dx} \quad (73c)$$

$$i=n+2: \quad \sum_{j=1}^n \frac{\bar{H}_j}{R_0 T_0} \frac{d \bar{\gamma}_j}{dx} + \left[\sum_{j=1}^n \bar{\gamma}_j \frac{\bar{C}_{pj}}{R_0} \right] \frac{d(\bar{T}/T_0)}{dx} - \frac{u^2}{u_s^2} \cdot \frac{u_s^2}{R_0 T_0} \frac{d \ln \bar{\rho}}{dx} = \frac{u^2}{u_s^2} \cdot \frac{u_s^2}{R_0 T_0} \cdot \frac{d \ln A}{dx} \quad (73d)$$

Equation (73a) is I(363) with the meanings of the indices interchanged; equation (73b) is I(367) with the signs of all terms reversed and the index replacements $j \rightarrow i$, $l \rightarrow j$; equation (73c) is I(299) applied to the equilibrium flow; and (73d) is I(293) applied to the equilibrium flow. Note that

$$\frac{\bar{H}_i}{R_0 T_0} = SHJ(i) \quad (74a)$$

$$\frac{\bar{T}}{T_0} = CT \quad (74b)$$

$$\left(\frac{\bar{u}}{u_s}\right)^2 = SU2 \quad (74c)$$

$$\frac{u_s^2}{R_0 T_0} = \frac{R_0 T_0}{W_0} \cdot \frac{1}{R_0 T_0} = \frac{1}{W_0} \quad (74d)$$

Equation I(296) is used in the coding of the quantity $1/W$ in (73c).

The coefficients A_{ij} for these equations (73) are loaded into the array $AAA(I,J)$ for $I = 1$ to $n + 2$ and $J = 1$ to $n + 2$. The constants B_i are loaded into $AAA(I, n + 3)$ for $I = 1$ to $n + 2$. The system of equations is then solved by calling subroutine $DLSOL$, and the resulting values for the derivatives (72) are retrieved from the $(n + 3)$ th column of AAA .

3.19.2 Derivatives for the Numerical Integration

During the numerical integration of the rate equations, the derivatives of the species concentrations are given explicitly by equation I(287). These derivatives are computed in the nest of $D\emptyset$ loops ending at statement 120; they are denoted by $DGJ(J)$.

The nature of the remaining derivatives to be determined depends upon the region of solution (upstream or downstream) and the type of gas model being used. The unknowns solved for in the various cases are summarized in Table I. When an electronic nonequilibrium model is being used (NT=2), there is a fourth dependent variable in the integration, the total enthalpy h_0 . However, its derivative is given explicitly by equation I(333), so that it is not involved in the simultaneous solution for the derivatives listed in Table I.

For a given type of gas model, the same set of equations is solved regardless of whether the solution is in the upstream region (inverse method) or the downstream region (direct integration). However, the matrix of coefficients and the vector of constants differ in the two cases because of the difference between the sets of unknowns being solved for. Figure 8 is a flowchart of the solution for the derivatives. By following this flowchart, one can easily write out the equations being solved in each of the four cases listed in Table I, and verify that they are identical with the text equations listed in the table. It should be noted that $NE = NT + 1$ is the number of unknowns and equations in each case. For $NT = 2$, the variable SCPGH contains the sum $\sum_{j=2}^n C_{pj} \gamma_j$, but for $NT = 1$ the sum is over all species (see subroutine CØMM).

3.20 Function EXP

The operating system on the IBM 360/75 at Avco produces several lines of diagnostic output whenever the Fortran function DEXP(X) underflows to give a value 0. To avoid such interruptions of the normal output, the IBM 360 version of NATA contains a Fortran-coded exponential function, EXP(X), which checks the argument X and sets the result to zero for $X \leq -180$. For $X > -180$, EXP calls DEXP to compute EXP. This routine is not used in the UNIVAC version of NATA because the UNIVAC system EXEC II does not produce a diagnostic message for exponential underflow.

TABLE I

DERIVATIVES DETERMINED BY EXACT DURING THE NUMERICAL INTEGRATION

Region	IUPD	Model Type	NT	Unknown Derivatives	Equations Solved
Upstream	1	Chemical	1	$\frac{d(T/T_0)}{dx}$, $\frac{d \ln \bar{A}}{dx}$	I(293), I(299)
Upstream	1	Electronic	2	$\frac{d(T/T_0)}{dx}$, $\frac{d \ln \bar{A}}{dx}$, $\frac{d(T_e/T_0)}{dx}$	I(328), I(331), I(332)
Downstream	0	Chemical	1	$\frac{d(T/T_0)}{dx}$, $\frac{d \ln \rho}{dx}$	I(293), I(299)
Downstream	0	Electronic	2	$\frac{d(T/T_0)}{dx}$, $\frac{d \ln \rho}{dx}$, $\frac{d(T_e/T_0)}{dx}$	I(328), I(331), I(332)

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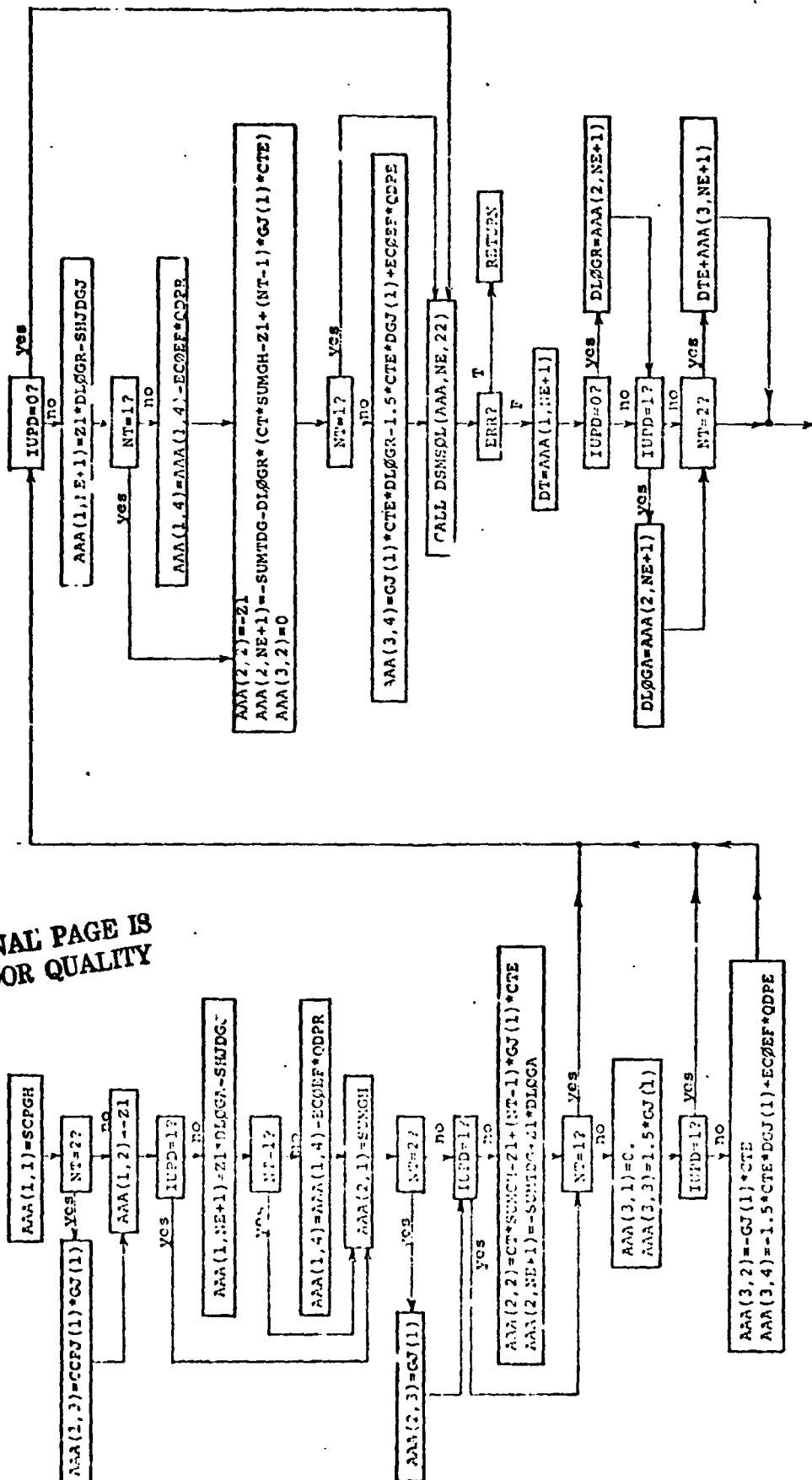


Figure 8. Flowchart of the Solution for the Derivatives of the Flow Variables in the Numerical Integration

3.21 Subroutine FINDX

The purpose of this subroutine is to determine the value of x at which the geometric area ratio has a given value A . An indicator (UPDOWN) specifies whether the upstream or the downstream solution is desired. An entry point FINDXC(A,MBL,X) permits determining the value of x at which a rectangular channel has a half-width A , in centimeters, on its MBLth profile.

The solution is obtained by a Newton-Raphson iterative procedure. The independent variable in the iteration is taken to be $V = |x|$.

If the area ratio is given (normal entry point), then its value is tested before the solution is attempted. If A is less than 0.99999, the case is terminated by calling DUMP, because by its definition the area ratio cannot be less than unity. If A is between 0.99999 and 1.0, the discrepancy is assumed to be due to a numerical (e.g., round-off) error elsewhere in the program; in this case, X is set to 0.0 (the value for $A = 1.$), and the RETURN is executed.

If A is greater than 1.0, V is initialized to 1.0 and the Newton-Raphson procedure is started. The geometric area ratio AR and its derivative $DADX$ at the point $X = V \cdot \text{sign}(\text{UPDOWN})$ are calculated by calling GEOMAR. If AR agrees with A to within one part in 10^5 , the iteration has converged. If not, a new estimate of V is calculated from the Newton-Raphson formula

$$V = V_0 + \frac{A-AR}{|DADX|} \quad (75)$$

However, V is not allowed to increase by more than a factor of 2 in any step; this is to prevent V from becoming very large after a step in which V was very close to the throat (where $DADX = 0.$). A maximum of 50 iterative steps is allowed.

If the channel width is given (entry FINDXC), then V is initialized to 100 cm, a value typical of axial positions where test panels might be installed in a channel. For each trial value of V, the profile half width $AR = Z(MBL)$ and slope $DADX = DZDX(MBL)$ are computed by calling the entry point GMAR3 of GEØMAR. The same Newton-Raphson formulas and convergence test are used as in the area ratio case.

3.22 Subroutine FRØZEQ

FRØZEQ is the routine which controls the computation of the frozen and equilibrium flow solutions. The frozen flow solution is generated when the routine is called through entry FRØZEN. Entry EQUIL(IPASS) gives the equilibrium solution.

3.22. Nondimensionalization of the Flow Variables

Throughout subroutine FRØZEQ, the calculations are carried out with most of the flow variables in nondimensional form. The relation between the Fortran symbols and the dimensional flow variables is summarized in the following formulas:

AFNX	A_e , effective area ratio
CH	$\frac{W_0 h}{R_0 T_0}$, nondimensional specific enthalpy
CHA	$\frac{W_0 h_0}{R_0 T_0}$, nondimensional enthalpy in the reservoir
CT	T/T_0 , nondimensional temperature
CX	x, axial nozzle coordinate (cm)
FLUX	$\frac{\rho u}{\rho_0 u_s}$, nondimensional mass flux

PRES	$\frac{p}{p_0}$, nondimensional pressure
RHØ	$\frac{\rho}{\rho_0}$, nondimensional density
SU	$\frac{u}{u_s}$, nondimensional velocity

In these definitions, the subscript 0 denotes conditions in the reservoir, R_0 is the universal gas constant, and u_s is a velocity defined by

$$u_s = \sqrt{R_0 T_0 / W_0} \quad (76)$$

where W_0 is the molecular weight in the reservoir and R_0 is the universal gas constant in mechanical cgs units (erg/mole °K).

3.22.2 Determination of Frozen Sonic Conditions

When entry FRØZEN is called, subroutine FRØZEQ determines the sonic conditions in the frozen flow before starting the main frozen flow solution.* This calculation is done by the statements from entry FRØZEN down to entry EQUIL. A flowchart of this part of the subroutine is shown in figure 9. The criterion for locating the sonic point is that the mass flux m be a maximum. This condition follows from the continuity equation I(243) and the fact that the flow area A is a minimum at the sonic throat. The temperature at the sonic point is determined as follows: the temperature is initialized to its reservoir value, and is then decremented repeatedly by the amount $DEL T_1 \cdot T_0$, where $DEL T_1$ is normally 0.01. At each of these temperatures, the corresponding mass flux in frozen flow is calculated by calling subroutines THERM and PRØP. This procedure is continued until the mass flux passes its maximum. Then the temperature step is reduced and an iteration is carried out to locate the sonic-point temperature with high accuracy.

*The calculation of sonic conditions for the equilibrium solution is done in a separate subroutine, NRMAX.

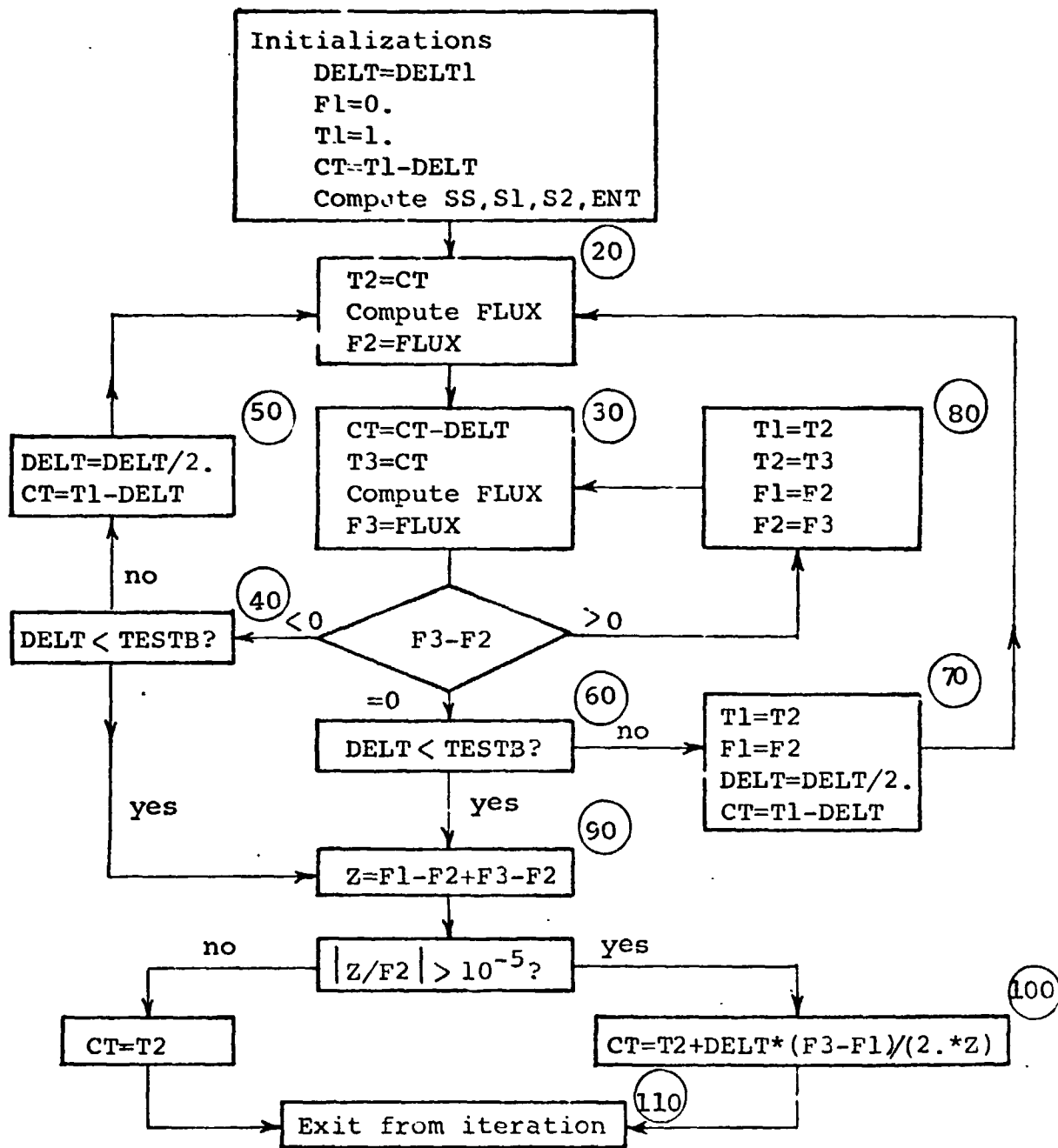
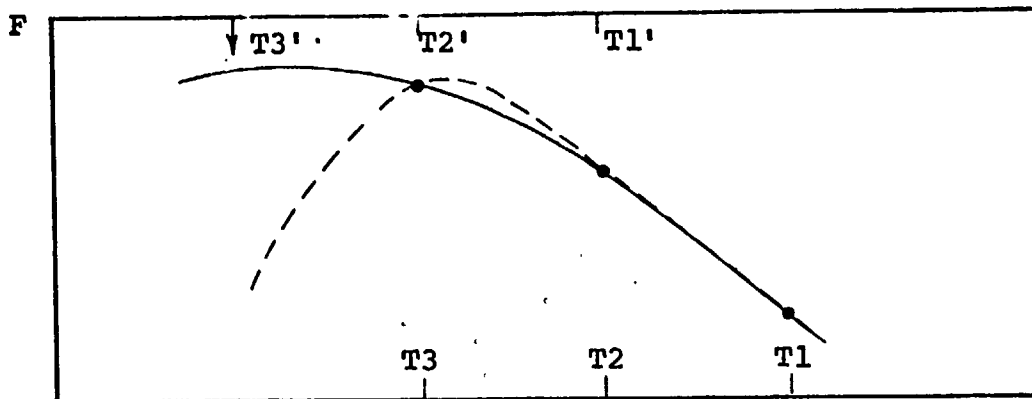


Figure 9. Flow Chart for Determination of Sonic Conditions

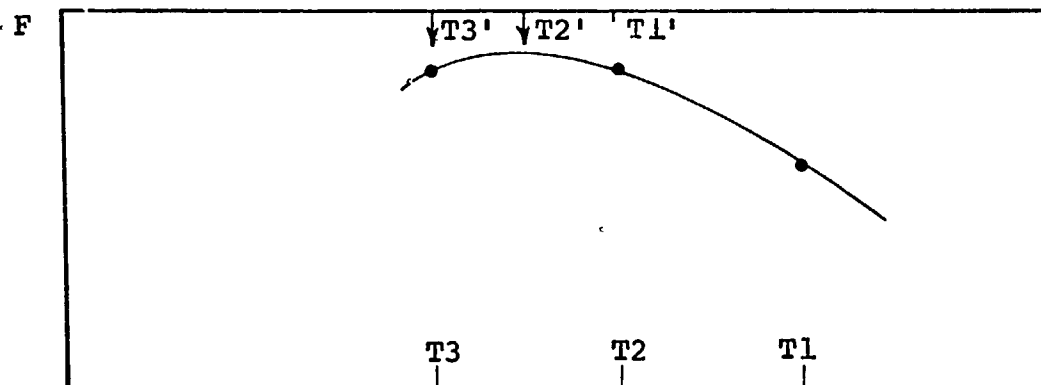
The algorithm for determining the sonic-point temperature (figure 9) is based upon consideration of a sequence of three equally-spaced temperatures $T_1 > T_2 > T_3$ and the corresponding mass flux values F_1, F_2, F_3 . T_1 and T_2 are always chosen in such a way that $F_2 > F_1$; thus, T_1 is definitely known to be higher than the sonic temperature. At each stage of the iteration, the rules for selecting the next triplet of temperatures T_1', T_2', T_3' depend upon the value of $F_3 - F_2$. There are three cases, which are illustrated schematically in figure 10. In case A ($F_3 > F_2$), the three mass flux values form a monotonic sequence $F_3 > F_2 > F_1$. In this case, the available data provide no indication that the point of maximum mass flux has been reached, although this may actually be the case, as shown by the dashed curve in the top section of the figure. In case A, T_1' is set equal to T_2 , T_2' to T_3 , and T_3' is obtained by applying the same decrement $T_1' - T_2'$ to T_2' . In case B ($F_3 = F_2$), the sonic point is known to lie between T_2 and T_3 . In this case, T_1' is set equal to T_2 , the temperature decrement $DELTA = \Delta T$ is cut in half, and the remaining two temperatures are set to $T_2' = T_1' - \Delta T$ and $T_3' = T_2' - \Delta T = T_3$. In case C ($F_3 < F_2$), the sonic point could lie between T_1 and T_2 or between T_2 and T_3 . To ensure that T_1' lies above the sonic temperature, it is set equal to T_1 . The decrement ΔT is halved as in case B, and $T_2' = T_1' - \Delta T$, $T_3' = T_2' - \Delta T = T_2$. If the sonic point actually lies between T_2 and T_3 , as illustrated by the solid curve in the bottom part of figure 10, the situation in the next step of the iteration will be as in case A.

At each step, the magnitude of $DELTA$ is compared with the criterion value $TESTB = 10^{-5}$. If $DELTA < TESTB$, the iteration has converged. A final improvement in the value of the sonic temperature is then obtained by fitting a parabola to the mass flux versus temperature data for the final triplet of points, and determining the maximum analytically based on this quadratic relation. The form assumed is

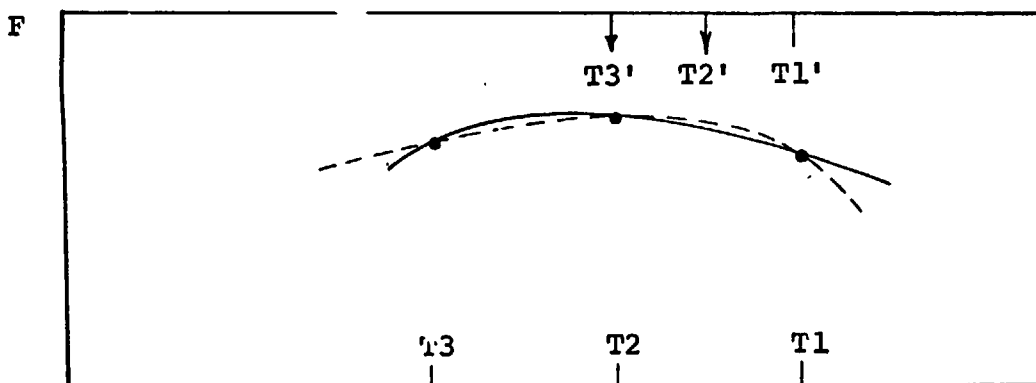
$$F(T) = A + B(T - T_2) + C(T - T_2)^2 \quad (77)$$



Case A: $F_3 > F_2$



Case B: $F_3 = F_2$



Case C: $F_3 < F_2$

Figure 10. Three Cases in the Algorithm for Finding the Throat Temperature

Substitution of the data points (T_1, F_1) , (T_2, F_2) , and (T_3, F_3) and solution for the coefficients A, B, C give

$$F = F_2 - \frac{F_3 - F_1}{2(\Delta T)} (T - T_2) + \frac{F_1 + F_3 - 2F_2}{2(\Delta T)^2} (T - T_2)^2 \quad (78)$$

Setting $dF/dT = 0$ then leads to the final result

$$T_* = T_2 + \frac{(F_3 - F_1)(\Delta T)}{2(F_1 + F_3 - 2F_2)} \quad (79)$$

If the quantity $Z/F_2 = (F_1 + F_3 - 2F_2)/F_2$ is less than or equal to 10^{-5} , equation (77) is not used. In this case, T_* is simply set equal to T_2 , to avoid possible errors due to loss of accuracy on the subtraction in Z (or a divide check in (79) if $Z = 0$).

3.22.3 Main Flow Solution

The basic operation performed by subroutine FRØZEQ is the calculation of the frozen and equilibrium flow solutions. In its essentials, this calculation is quite simple. The nondimensional temperature CT is decremented repeatedly, starting from its reservoir value (CT = 1.). At each value of the temperature, the other flow variables are computed, using subroutines THERM and PRØP in the case of frozen flow and subroutine NEWRAP in the case of equilibrium flow. Then the position x in the nozzle at which the flow point occurs is determined from the known nozzle geometry and the value of the area ratio at the flow point, and the results are printed by calling entry ØUT2 of subroutine ØUT1.

The actual complexity of FRØZEQ, as illustrated by its flowchart (figure 11) is due to provisions for treating various options and regions of the solution. The control variables pertaining to some of these cases and conditions will be defined before the subroutine is discussed in detail:

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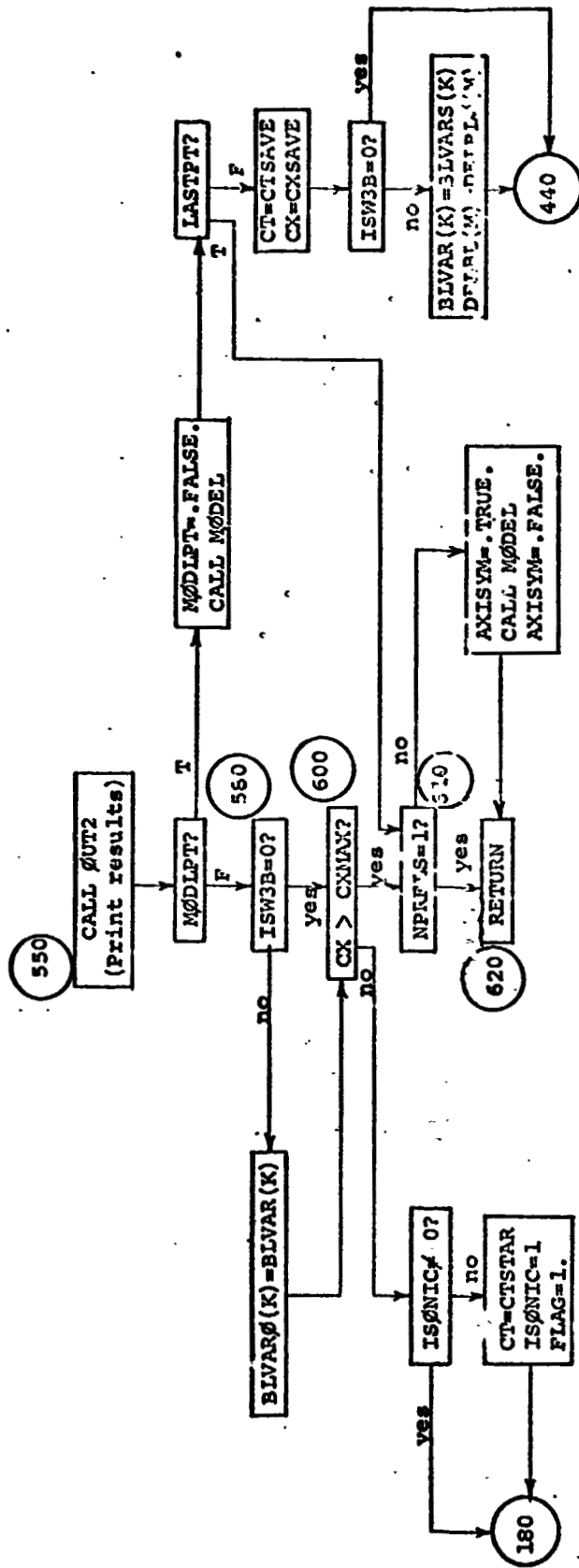


Figure 11d. Flowchart of subroutine FRØZEQ (Part d)

ISW3B = 0	Boundary layer is omitted.
ISW3B ≠ 0	Boundary layer is included.
IPASS = 1	Preliminary equilibrium solution from the reservoir to the sonic point, to determine the boundary layer displacement thickness at the throat for use in reservoir condition calculations.
IPASS = 2	Final flow solution.
ISØNIC = -1	Region upstream of the throat.
ISØNIC = 0	Sonic point.
ISØNIC = 1	Region downstream of the throat.
MØDLPT = .FALSE.	Regular point of the flow solution.
MØDLPT = .TRUE.	Iteration to find the conditions at a specified model point is in progress.
SKIP = .FALSE.	Regular flow point or model point.
SKIP = .TRUE.	Regular flow point just after a model point.
LASTPT = .FALSE.	Regular flow point or model point.
LASTPT = .TRUE.	Last point of solution.

The simplest case is that of a solution without the boundary layer and without any model points. In this case, FRØZ EQ generates a point of the solution by traversing the loop outlined by the statement numbers 180, 190, 220, 310, 410, 420, 440, 450, 550, 580, 600. The temperature decrement DELTV is equal to the input value of DELT1 (normally 0.01) in this case, until the difference $XØ - XP$ in spatial position between two successive flow points exceeds 10 cm. Once this has happened, the unnumbered statements between 180 and 190 reduce DELTV so as to maintain a spatial step size of approximately 10 cm. This variation of DELTV occurs far downstream of the throat, in the region where the free stream temperature has fallen to a small fraction of the reservoir

temperature. DELTV is not allowed to fall below 0.02 DELT1. These adjustments of DELTV in the region far downstream are also made when the boundary layer is included in the solution.

The second statement below 190 tests whether the temperature CT is below the sonic temperature CTMAX, i.e., whether the solution has passed the throat. If it has, in the case without the boundary layer, ISØNIC is reset from -1 to +1, and the indicator FLAG for upstream and downstream solutions in the geometry routines is reset from -1. to +1. Once the flow conditions for the temperature CT have been computed, the spatial position in the nozzle is determined by calling subroutine FINDX(AFNX, FLAG, CX) (statement 310). This subroutine determines the upstream (FLAG = -1.) or downstream (FLAG = 1.) position CX at which the geometric area ratio is equal to AFNX.

If the boundary layer is included, the input value of DELT1 is saved in the location DELTSV, and DELT1 is reset to $0.049(1. - CTMAX)$ during initialization. This ensures that there will be 20 flow points upstream of the throat. In some cases with $CTMAX \sim 0.9$, the normal input value of $DELT1 = 0.01$ would give only about 10 points above the throat, which would adversely affect the accuracy of the numerical integration in the boundary layer routine BLAYER. Once the throat has been reached, DELT1 is restored to its original value by the statements below 320.

When the boundary layer is included, the effective area ratio $A_e = AFNX$ is no longer equal to the geometric area ratio $A_g = AR$, because of the boundary layer displacement thickness, δ^* . The relation between these two area ratios (Section 4.4 of Volume I) involves the displacement thickness at the throat, δ^* . Since this is not known initially, the spatial position CX corresponding to each flow point is calculated on the assumption that $A_g \simeq A_e$ until the throat has been reached. These calculations are done using subroutine FINDX, as in the case without the boundary layer. A special flow point is computed at the throat (ISØNIC = 0, CT = CTSTAR, CX = 0.) in order to determine the quantity

$1 - \delta_*^* = \text{ØMDST}$ (statement 390) for the nozzle (or two such values in the case of a channel). Then, beginning at the third point beyond the throat, CX is computed taking the displacement thickness into account, using subroutine AGSØLN. In statement 340, AGSØLN is called with the effective area ratio AFNX for the current step and the displacement thickness DELBL for the preceding step. The resulting CX value is used in the boundary layer calculations (statement below 370). Subroutine AGSØLN is then called again, with the same AFNX and the new DELBL produced by the call to BLAYER. This second call to ACSØLN gives an improved value of CX.

In some problems, at large distances downstream of the throat, the displacement thickness becomes large enough to have a major effect on the effective area ratio. Under such circumstances, the coupling between the boundary layer and the inviscid flow solution can lead to computational instability. The boundary layer routine BLAYER contains a smoothing algorithm intended to suppress such instability (Section 5.11 of Volume I). This algorithm involves a parameter W, preset to a value WSAVE which usually prevents instability. However, occasionally instability develops in spite of the smoothing. In the frozen and equilibrium solutions, the symptom of such instability is that CX begins to decrease with decreasing temperature. When this occurs, subroutine FRØZEQ attempts to produce a valid solution by cutting the stabilizing parameter W in half and restarting the solution in the reservoir. This operation is accomplished by coding between statements 340 and 350. If instability has been encountered after three successive restarts, the solution is abandoned and error dumps are written. Also, restarting is not attempted if the geometric area ratio is more than 4 times greater than the effective area ratio at the point where CX decreases, as in this case the code is probably predicting actual choking of the flow by boundary layer growth.

3.22.4 Model Points

Model points are input-specified spatial locations, in the downstream region of the nozzle, at which special calculations are to be performed. Since model points are

specified by values of CX, while points of the equilibrium or frozen flow solution are specified by values of the temperature, an iteration is required to find the temperature CT corresponding to each model point.

During the initialization, and after the completion of each model point calculation, subroutine NEXTMP is called to determine the location XMØDEL of the next model point. During the main flow solution, after CX has been determined for each flow point, statement 450 tests whether CX is greater than or equal to XMØDEL. If not, the flow solution proceeds normally. If $CX \geq XMØDEL$, the iteration to determine the CT value corresponding to $CX = XMØDEL$ is started. While this iteration is in progress, the indicator MØDLPT is .TRUE..

To avoid complicating the programming of subroutine BLAYER, and to save computer time, BLAYER is not called during the model point iteration. Instead, the boundary layer properties at the model point are approximated by linear interpolation of the data at the preceding and current main flow points, and at each cycle of the iteration subroutine TRANSP is called (statement 360) to determine the transport properties at the current temperature.

The iterative algorithm for improving the estimate of CT corresponding to XMØDEL is a numerical Newton-Raphson technique embodied in statement 490:

$$T = T_1 + (T_2 - T_1) \cdot \frac{X_m - X_1}{X_2 - X_1} \quad (80)$$

where T_2 is the temperature for which calculations were done in the current step, T_1 is the temperature in the preceding iterative step, X_2 and X_1 are the CX values corresponding to T_2 and T_1 , and X_m is XMØDEL.

The convergence criterion for the iteration is that CX differ from XMØDEL by no more than 0.0025 cm or 0.001 inch. A maximum of 10 iterations is allowed. If convergence has not been achieved after 10 iterations, a diagnostic message

is written and the data from the final iteration are used. When the iteration has thus been terminated by either convergence or nonconvergence, the results for the flow conditions at the model point are printed by calling `ØUT2`, `MØDLPT` is reset to `.FALSE.`, and subroutine `MØDEL` is called to perform the required calculations of conditions on models. Also, the indicator `SKIP` is set to `.TRUE.`, and the next `XMØDEL` is determined by calling `NEXTMP`. Before resuming the main flow solution, `FRØZEQ` checks whether the next `XMØDEL` is less than or equal to the `CX` for the current main solution point. If so, the model point iteration is started again to find the flow conditions at the new `XMØDEL`.

Once all of the `XMØDEL` values less than or equal to the `CX` for the current main flow point have been dealt with, the flow conditions for the current flow point are recomputed and printed. The `.TRUE.` value of the indicator `SKIP` provides for this recomputation by causing a transfer to statement 220 instead of 550 just below statement 450. Also for `SKIP = .TRUE.`, the determination of `CX` and the call to `BLAYER` are bypassed below 220. `SKIP` is then reset to `.FALSE.`, the results are printed by calling `ØUT2`, and the main flow solution is resumed.

When a `CX` greater than or equal to `CXMAX` is reached, `XMØDEL` is set equal to `CXMAX` and the indicator `LASTPT` is set to `.TRUE.`. The model point iteration then provides the flow conditions at the position `CXMAX`. After the iteration has converged and the results have been printed, for `LASTPT = .TRUE.`, control is transferred to statement 610. For nozzle flow problems (`NPRFLS = 1`), the `RETURN` is then executed. For channel problems (`NPRFLS = 2`), subroutine `MØDEL` is first called with `AXISYM = .TRUE.` to provide calculation of the stagnation conditions in the channel exit plane. Stagnation condition calculations are suppressed for channels, with this one exception, because stagnation point models and Pitot probes are not used inside channels.

3.23 Subroutine GEØM

The function of `GEØM` is to calculate the density, the effective area ratio and the derivative of one or the other of these quantities at each position `x` along the nozzle during

the nonequilibrium flow solution. Two different methods of calculation are used in different parts of the flow solution. During the solution by the perturbation method, and during the direct integration of the rate equations in the region beyond the call to THRØAT, the calculation of the effective area ratio is based directly on the nozzle geometry and the boundary layer displacement thickness, if any, and the density is then calculated from the continuity equation. During the numerical integration by the inverse method upstream of the call to THRØAT, the density is obtained from the analytical density-area relation I(383):

$$\left(\frac{\rho}{\rho_0} A_e\right)^2 \left[1 - \left(\frac{\rho}{\rho_0}\right)^\alpha\right] = C \quad (81)$$

and the effective area ratio is then obtained from the continuity equation I(266). The constants α and C in (81) are determined in the main program from sonic-point conditions, prior to the first call to GEØM, as explained in Section 7.4 of Volume I and in Section 3.1 of the present report.

The overall logical structure of GEØM is diagrammed in figure 12. Immediately upon entry, the subroutine calls GEØMAR to compute the geometric area ratio $S1$ and its derivative $S2$ at the current position CX . If the nonequilibrium flow solution being computed includes the boundary layer ($ISW3B \neq 0$), then AESØLN is called to convert $S1$ into the effective area ratio and $S2$ into the derivative of the effective area ratio, based on the current displacement thickness.

If the flow solution is beyond the call to THRØAT ($IUPD = 0$) or if the perturbation technique is in use ($INEQ = 0$), the effective area ratio $AFNX$ is set equal to $S1$ and the nondimensional density $RHØ$ is calculated from $AFNX$ using the continuity equation. Also, the derivative $DLØGA = d \ln A_e / dx$ is computed from $S1$ and $S2$.

On the other hand, if the numerical integration technique and the inverse method are being used to generate the flow solution ($INEQ = 1$ and $IUPD = 1$), then the nondimensional

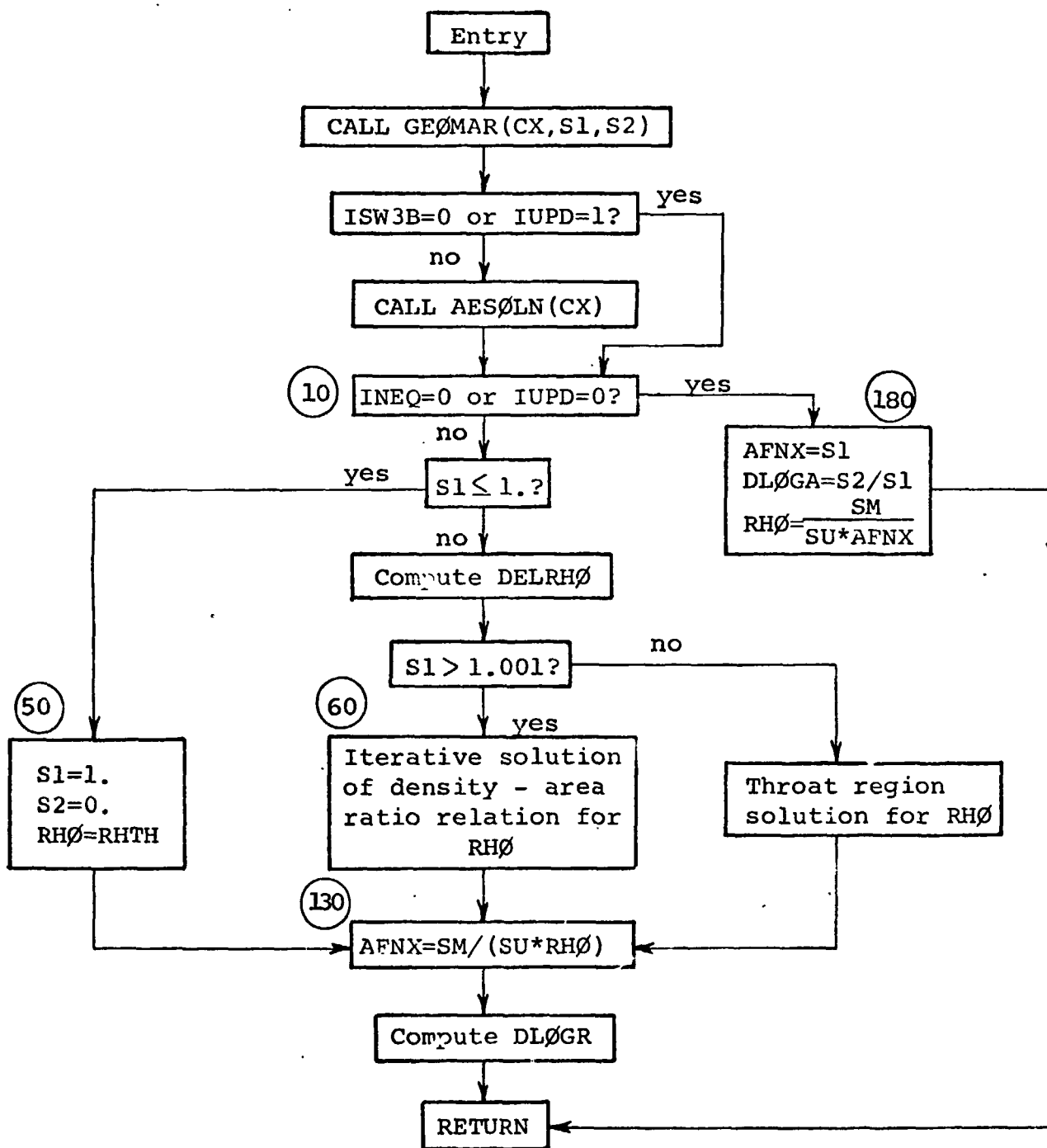


Figure 12. Flowchart of Subroutine GEØM

density $RH\emptyset$ is calculated from S1 by solving the density-area ratio relation (81), and the effective area ratio $AFNX$ is recomputed from $RH\emptyset$ using the continuity equation. The derivative $DL\emptysetGR = d \ln(\rho/\rho_0)/dx$ is then computed for use in the rate equations (see the discussion of subroutine EXACT above).

The calculation of $RH\emptyset$ from equation (81) is based on the following analysis. Equation (81) can be solved for A_e :

$$A_e = \frac{\rho_0}{\rho} \sqrt{\frac{C}{1 - (\rho/\rho_0)^\alpha}} \quad (82)$$

The form of this relation is illustrated in figure 13, for the case $\alpha = 0.2$, $C = 0.035049$. The corresponding density ratio at the sonic point is $\rho^*/\rho_0 = 0.62092$. The basic computational problem in GEOM is to solve (81) or (82) for ρ/ρ_0 , based on a given value of A_e . Except near the throat, the solution is carried out using the Newton-Raphson method; i.e., from a given estimate ρ_p of ρ , an improved estimate ρ_n is obtained as follows:

$$A_p = \frac{\rho_0}{\rho_p} \sqrt{\frac{C}{1 - (\rho_p/\rho_0)^\alpha}} \quad (83a)$$

$$\rho_0 \left(\frac{dA}{d\rho} \right)_p = \frac{\rho_0 A_p}{\rho_p} \cdot \frac{(1 + \frac{1}{2}\alpha) (\rho_p/\rho_0)^\alpha - 1}{1 - (\rho_p/\rho_0)^\alpha} \quad (83b)$$

$$\rho_n = \rho_p + \frac{A_e - A_p}{(dA/d\rho)_p} \quad (83c)$$

Here A_p is the A_e value calculated from (82) for $\rho = \rho_p$; $(dA/d\rho)_p$ is the derivative of (82) evaluated at $\rho = \rho_p$; and equation (83c) is the Newton-Raphson formula.

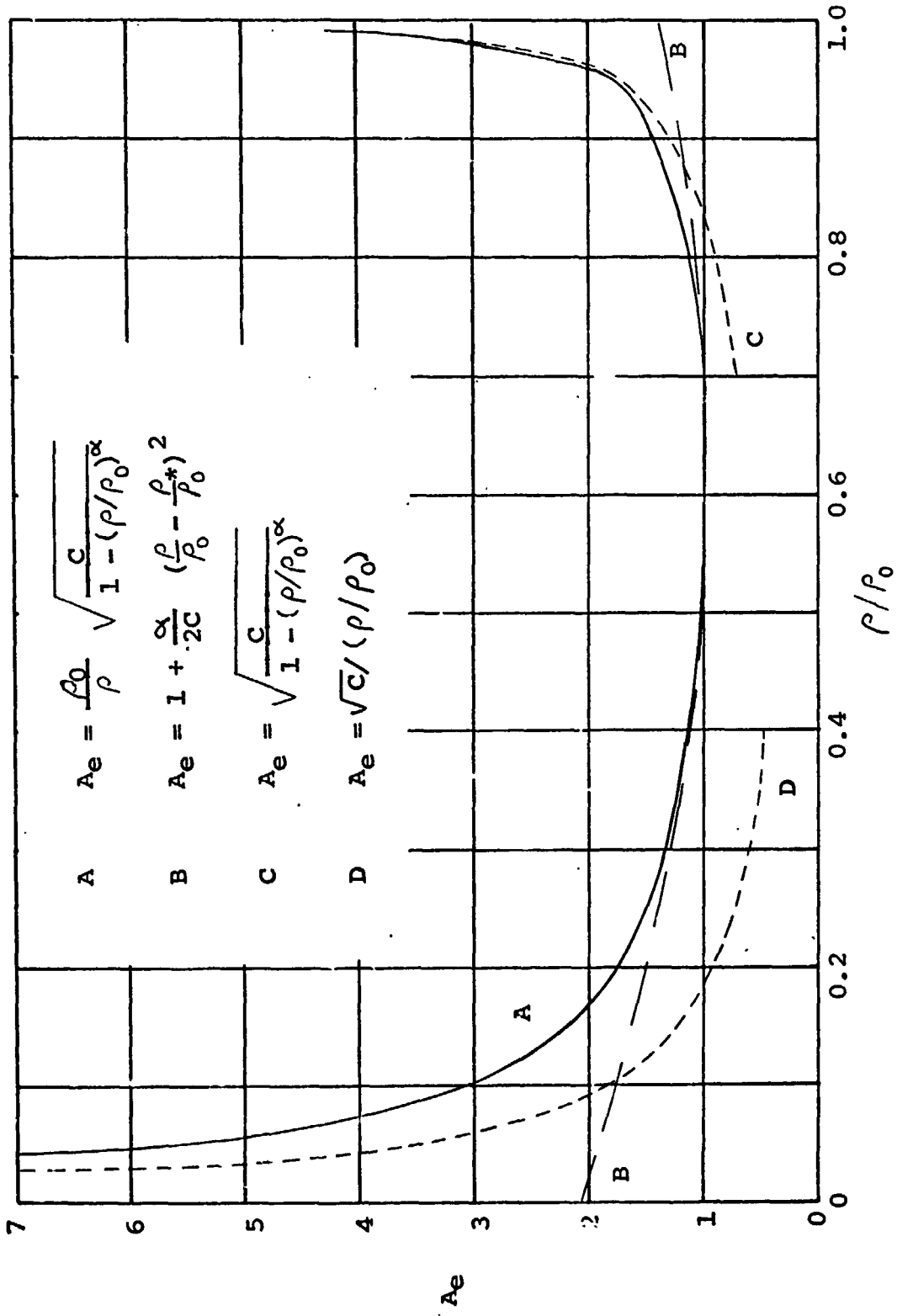


Figure 13. Approximations to the Density-Area Ratio Relation

Examination of equations (83) reveals several potential problems. At the sonic point ($A_e = 1$), the derivative $dA_e/d\rho$ is zero, so that the second term in (83c) is undefined. Near the throat, $(dA/d\rho)_p$ is very small. If a trial value ρ_p is near the throat, and nearer the throat than the solution of (82), the Newton-Raphson formula (83c) can give an "improved" ρ_n which is much less accurate than ρ_p and which may even be outside the permissible range $0 \leq \rho \leq 1$. To deal with these problems reliably, GEOM uses the following algorithm:

- (1) If $A_e \leq 1$, $\rho = \rho_*$ (throat point)
- (2) If $1 < A_e \leq 1.001$ (throat region), ρ is calculated from a Taylor series expansion of (82) around the throat point:

$$\begin{aligned}
 A_e &= 1 + \frac{1}{2} \left(\frac{d^2 A_e}{d\rho^2} \right)_{\rho=\rho_*} (\rho - \rho_*)^2 + \frac{1}{6} \left(\frac{d^3 A_e}{d\rho^3} \right)_{\rho=\rho_*} (\rho - \rho_*)^3 \\
 &= 1 + \frac{\alpha}{2C} \left(\frac{\rho}{\rho_0} - \frac{\rho_*}{\rho_0} \right)^2 + \frac{(\alpha+1)\alpha^{3/2}}{6C\sqrt{C(\alpha+2)}} \left(\frac{\rho}{\rho_0} - \frac{\rho_*}{\rho_0} \right)^3
 \end{aligned}
 \tag{84}$$

The coefficients in this equation have been evaluated by differentiation of (82) and use of the sonic-point conditions $\rho = \rho_*$, $A_e = 1$, $(dA_e/d\rho)_* = 0$. In the throat region, where (84) is used, the cubic term is small. If this term were neglected, the solution of (84) would be

$$\frac{\rho}{\rho_0} - \frac{\rho_*}{\rho_0} = s \sqrt{\frac{2C}{\alpha}} (A_e - 1) \equiv s \cdot (\delta\rho) \tag{85}$$

where S is $+1$ upstream and -1 downstream of the throat. When the cubic term is taken into account, the solution of (84) can be approximated by

$$\frac{\rho}{\rho_0} - \frac{\rho_*}{\rho_0} = s \left[\frac{2C(A_e - 1)}{\alpha} - \frac{S(\alpha + 1)}{3} \sqrt{\frac{\alpha}{C(\alpha + 2)}} \cdot (\delta\rho)^3 \right]^{1/2} \quad (86)$$

where $(\delta\rho)$ is defined by the second equality in (85). Equation (86) is used to calculate ρ/ρ_0 when $A_e < 1.001$.

- (3) If $A_e > 1.001$ (general case), the Newton-Raphson procedure (83) is used. To ensure its stability, pains are taken to select a first estimate of ρ which is farther from ρ_* than the solution point. Since the relation (82) is everywhere concave upward (figure 13), the Newton-Raphson formula undercorrects ρ whenever the estimate ρ_p lies too far from ρ_* . Thus, if the first estimate ρ_p is too far from ρ_* the iteration converges to the solution without ever overshooting it. Under these circumstances, no instability can develop. To obtain such an initial estimate, the subroutine uses three approximations to (82), all of which lie below (82). One of these approximations is (85). The second is an approximation for $\rho/\rho_0 \sim 1$, namely

$$A_e \approx \sqrt{\frac{C}{1 - (\rho/\rho_0)^\alpha}}$$

from which

$$\frac{\rho}{\rho_0} \approx \left(1 - \frac{C}{A_e^2} \right)^{1/\alpha} \quad (87)$$

The third is an approximation for $\rho \sim 0$,

$$A_e \approx \sqrt{C}/(\rho/\rho_0)$$

from which

$$\frac{\rho}{\rho_0} \approx \sqrt{C}/A_e \quad (88)$$

In the upstream region ($dA_e/dx < 0$), the initial estimate is chosen to be

$$\frac{\rho_p}{\rho_0} = \min \left[\frac{\rho_*}{\rho_0} + \sqrt{\frac{2C(A_e - 1)}{\alpha}}, \left(1 - \frac{C}{A_e^2} \right)^{1/\alpha} \right] \quad (89)$$

In the downstream region ($dA_e/dx > 0$), it is

$$\frac{\rho_p}{\rho_0} = \max \left[\frac{\rho_*}{\rho_0} - \sqrt{\frac{2C(A_e - 1)}{\alpha}}, \frac{\sqrt{C}}{A_e} \right] \quad (90)$$

Once the solution for ρ has been obtained, the derivative $d \ln \rho / dx$ is calculated. Normally this is determined from the formula

$$\frac{d \ln \rho}{dx} = \frac{2C}{\alpha \left(\frac{\rho A_e}{\rho_0} \right)^2 - (\alpha + 2)C} \cdot \frac{d \ln A_e}{dx} \quad (91)$$

which can be obtained by differentiating (82). The factor $d \ln A_e / dx$ on the right is evaluated as S_2/S_1 , based on the specified nozzle geometry and displacement thickness. The throat point ($A_e = 1$) is a special case, because there the denominator of (91) is zero and $dA_e/dx = 0$, so that (91) is of the form $0/0$. At the throat, $d \ln \rho / dx$ must be evaluated from (91) using L'Hospital's rule:

$$\left(\frac{d \ln \rho}{dx} \right)_{A_e=1} = 2C \lim_{A_e \rightarrow 1} \frac{d^2 \ln A_e / dx^2}{\frac{2\alpha}{\rho_0} \left(\rho \frac{dA_e}{dx} + A_e \frac{d\rho}{dx} \right)}$$

Since $(dA_e/dx)_* = 0$, this gives

$$\left(\frac{d \ln \rho}{dx} \right)_* = - \sqrt{\frac{C(d^2 \ln A_e / dx^2)_*}{\alpha \rho_* / \rho_0}} \quad (92)$$

The area derivative on the right hand side has to be evaluated from the specified nozzle geometry. In general,

$$\left(\frac{d^2 \ln A_e}{dx^2} \right)_* = \left(\frac{d^2 A_e}{dx^2} \right)_* \quad (93)$$

by virtue of the conditions $A_{e*} = 1$, $(dA_e/dx)_* = 0$. To proceed further, it is necessary to consider the three geometry options separately:

(1) Two-dimensional nozzle

In this case, from equation I(126) with y_* replaced by y_0 as usual,

$$\left(\frac{d^2 A_e}{dx^2}\right)_* = \frac{\frac{1}{y_0} \left(\frac{d^2 y}{dx^2}\right)_* - \frac{1}{y_0} \left(\frac{d^2 \delta^*}{dx^2}\right)_*}{1 - \delta^*/y_0} \quad (94)$$

The term $(d^2 \delta^*/dx^2)_*$ is small, because the displacement thickness δ^* is small and slowly varying at the throat. This term is therefore neglected. Since the throat sections of the profile fit are always of the form I(122) (shape 2, circular arc convex toward the axis), the remaining term in the numerator of (94) is given by

$$\frac{1}{y_0} \left(\frac{d^2 y}{dx^2}\right)_* = \frac{1}{y_0} \left(\frac{d^2 y}{dx^2}\right)_0 = \frac{1}{y_0 P_3}$$

The second equality is obtained by noting that $P_2 = 0$ for a throat section. Thus,

$$\left(\frac{d^2 A_e}{dx^2}\right)_* = \frac{1}{y_0 P_3 (1 - \delta^*/y_0)} \quad (95)$$

(2) Axisymmetric nozzle

From equation I(131) since $(dA_e/dx)_* = 0$ and $A_{e*} = 1$,

$$\left(\frac{d^2 A_e}{dx^2}\right)_* = \frac{1}{1 - \delta^*/y_0} \left[\frac{1}{\sqrt{A_{g*}}} \left(\frac{d^2 A_g}{dx^2}\right)_* - \frac{1}{2A_{g*}^{3/2}} \left(\frac{dA_g}{dx}\right)_*^2 - \frac{2}{y_0} \left(\frac{d^2 \delta^*}{dx^2}\right)_* \right] \quad (96)$$

Now dA_g/dx is very small at the sonic point (which is quite near the throat), so that the square of this quantity can be neglected. The quantity $(d^2 \delta^*/dx^2)_*$ is also neglected for reasons discussed above. Hence,

$$\left(\frac{d^2 A_e}{dx^2}\right)_* \simeq \frac{(d^2 A_g/dx^2)_*}{\sqrt{A_{g*}}(1 - \delta^*/y_0)} \quad (97)$$

Evaluation of $(d^2 A_g/dx^2)_*$ using I(116) and I(122) then gives

$$\left(\frac{d^2 A_e}{dx^2}\right)_* \simeq \frac{2}{y_0 P_3 (1 - \delta^*/y_0) \sqrt{A_{g*}}} \quad (98)$$

(3) Rectangular channel

From equation I(134),

$$\begin{aligned} \left(\frac{d^2 A_e}{dx^2}\right)_* = \frac{1}{D} \left[(y - \delta_1^*) \left(\frac{d^2 z}{dx^2} - \frac{d^2 \delta_2^*}{dx^2}\right) + (z - \delta_2^*) \left(\frac{d^2 y}{dx^2} - \frac{d^2 \delta_1^*}{dx^2}\right) \right. \\ \left. + 2 \left(\frac{dy}{dx} - \frac{d\delta_1^*}{dx}\right) \left(\frac{dz}{dx} - \frac{d\delta_2^*}{dx}\right) \right]_* \quad (99) \end{aligned}$$

where

$$D = (y_0 - \delta_{1*}^*)(z_0 - \delta_{2*}^*) \quad (100)$$

At the sonic point, the quantities $\left(\frac{dy}{dx} - \frac{d\delta_1^*}{dx}\right)$ and $\left(\frac{dz}{dx} - \frac{d\delta_2^*}{dx}\right)$ are small, so that their product is considered negligible. The quantities $(d^2 \delta_i^*/dx^2)_*$ are also neglected, as in the other options. Also, $y_* \simeq y_0$ and $z_* \simeq z_0$. Hence,

$$\left(\frac{d^2 A_e}{dx^2}\right)_* \approx \frac{(y_0 - \delta_{1*}^*) (d^2 z/dx^2)_* + (z_0 - \delta_{2*}^*) (d^2 y/dx^2)_*}{(y_0 - \delta_{1*}^*) (z_0 - \delta_{2*}^*)} \quad (101)$$

Evaluation of $(d^2 z/dx^2)_*$ and $(d^2 y/dx^2)_*$ from I(122) gives, finally,

$$\left(\frac{d^2 A}{dx^2}\right)_* \approx \frac{1}{(z_0 - \delta_{2*}^*) P_{3z}} + \frac{1}{(y_0 - \delta_{1*}^*) P_{3y}} \quad (102)$$

where P_{3z} is the P_3 parameter for the z profile and P_{3y} that for the y profile.

3.24 Subroutine GEØMAR

GEØMAR is the basic geometry routine in the NATA code. Through its various entries, it provides calculations of the geometric area ratio and its derivative and of profile ordinates and their derivatives both for nozzles and for rectangular channels.

The subroutine has 4 entries, with the following calls:

GEOMAR(X,ARATIØ, DERIVA)	(IENTRY = 1)
GMAR(X,Y)	(IENTRY = 2)
GMAR2(X,Y,Z)	(IENTRY = 3)
GMAR3(X,DYDX,DZDX,Y,Z)	(IENTRY = 4)

The arguments have the following meanings:

X	= axial coordinate in nozzle (cm)
ARATIØ	= geometric area ratio
DERIVA	= derivative of geometric area ratio (cm ⁻¹)
Y	= profile ordinate (cm)

Z = ordinate for second profile in a channel (cm)
 DYDX = derivative of profile ordinate
 DZDX = derivative of ordinate for second profile
 in a channel

For each profile (I) the subroutine first determines the profile section (J) in which the specified value of X lies. This is the first section for which $X < ATPI(J,I)$. The routine then calculates the profile ordinate from equation I(121), I(122), or I(123) depending on ISHAPE(J,I), and if required also calculates the derivative of the profile ordinate using the derivatives of these formulas:

(1) Straight Line, ISHAPE(J,I) = 1

$$\frac{dy}{dx} = P_2 \quad (103)$$

(2) Circular Arc Convex Downward, ISHAPE(J,I) = 2

$$\frac{dy}{dx} = \frac{x - P_2}{\sqrt{P_3^2 - (x - P_2)^2}} \quad (104)$$

(3) Circular Arc Concave Downward, ISHAPE(J,I) = 3

$$\frac{dy}{dx} = - \frac{x - P_2}{\sqrt{P_3^2 - (x - P_2)^2}} \quad (105)$$

The subroutine then calculates the geometric area ratio, if required, from equation I(118) for a two-dimensional nozzle (NPRFLS = 1, JDIM = 0), I(116) for an axisymmetric nozzle (NPRFLS = 1, JDIM = 1), or I(120) for a rectangular channel (NPRFLS = 2). It calculates the derivative of the area ratio from the derivatives of these formulas, i.e.,

$$\frac{dA_g}{dx} = \frac{1}{y_0} \frac{dy}{dx} \quad (2D \text{ nozzle}) \quad (106)$$

$$\frac{dA_g}{dx} = \frac{2y}{y_0^2} \frac{dy}{dx} \quad (\text{axisymmetric nozzle}) \quad (107)$$

$$\frac{dA_g}{dx} = \frac{y \frac{dz}{dx} + z \frac{dy}{dx}}{y_0 z_0} \quad (\text{channel}) \quad (108)$$

Before executing the RETURN, the routine rescales all of the computed results using the rescaling factor RSA. The area ratio and its derivative are multiplied by RSA in the case of a 2D nozzle, but in an axisymmetric nozzle or channel they are multiplied by $\sqrt{\text{RSA}}$. The factor RSA is initialized to 1.0 and is never changed in equilibrium and frozen flow solutions. In nonequilibrium flow solutions, RSA is reset in subroutine THRØAT so as to make the effective area ratio continuous between the upstream solution by the inverse method and the downstream direct solution.

3.25 Subroutine INGAS

This subroutine has two entry points, INGAS and INTA. The first part of the routine, which is entered by calling INGAS, sets up several arrays which are used in the chemical description of the gas mixture, as discussed in Section 2.1 of Volume I. The mathematical and Fortran notations for these arrays correspond as follows:

A_{ki}	AIN(K,I)	
$\bar{v}_{i-c,j}$	CDIJ(I,J)	(I = i - c)
$-1 + \nu_{i-c}^*$	BET(I)	(I = i - c)
q_i	QM(I)	
w_i	CGI(I)	

In the subroutine, the array AIN is first initialized to the square submatrix $\bar{\alpha}_{ij}$ of the matrix α_{ij} with $i = 1$ to c , where α_{ij} is the number of atoms of the j th element

per molecule of the i^{th} species, and c denotes the number of elements in the system. The inverse matrix A_{ki} of $\bar{\alpha}_{ij}$ is then computed and stored in AIN by calling subroutine MATINV.

Next, the matrix $\bar{\nu}$, which gives the composition of the dependent species in terms of the components, is then computed by evaluating equation I(6), and is stored in CDIJ. Then the singly subscripted array $-1 + \nu_{i-c}^*$, where ν^* is defined by equation I(15), is computed in BET. The array q_i giving the invariant composition of the gas in terms of the independent species is calculated using equations I(9) and I(10); it is denoted by QM(I). Finally, the species molecular weights $W_i = \text{CGI}(I)$ are computed as follows:

$$C_i = \sum_{j=1}^c \alpha_{ij} A_j \quad (109)$$

where A_j is the atomic weight of the j^{th} element;

$$W_i = C_i \quad (i = 1, \dots, c) \quad (110a)$$

$$W_i = \sum_{k=1}^c \bar{\nu}_{i-c,k} W_k \quad (i = c + 1, \dots, n) \quad (110b)$$

For the rationale of equation (110b), see equation I(5) and its associated discussion. The molecular weights of the dependent species $i = c + 1, \dots, n$ could also be calculated directly using equation (109), in place of (110b).

The purpose of the calculations following entry INTA is to compute the equilibrium conditions in the upstream reservoir. The actual thermochemical equilibrium calculation is performed by subroutine EQCALC(CTAP,PRESA). The argument CTAP is the reservoir temperature in degrees Kelvin, and PRESA is the reservoir pressure in atmospheres.

The results of the calculations are communicated from EQCALC to INGAS through the common blocks /EQC/ and /EQC2/.

3.26 Subroutine INIT

Subroutine INIT initializes a number of control parameters and nondimensionalizes the species thermochemical data using the reservoir temperature. Since the coding is straightforward, it will not be discussed.

3.27 Subroutine KANDMU

Subroutine KANDMU calculates quantities proportional to the viscosity and the translational thermal conductivity of the gas mixture from equations I(85) to I(87). The first section of the subroutine, down through statement 20, computes the quantities $(25/8)\sqrt{T} A_i^{(K)}$ and $(25/8)\sqrt{T} A_i^{(\mu)}$ defined by equation I(87), for all species i in the mixture. The value of $(25/8)\sqrt{T} A_i^{(\mu)}$ is obtained simply by dividing the factor ZM2(I) out of the quantity B(I,2) computed previously in subroutine KINT, equation (113) below, and is stored again in the array B(I,2), while the quantity $(25/8)\sqrt{T} A_i^{(K)}$ is computed directly from I(87) using the matrix elements Q(3,I,J) calculated previously in subroutine TRANSP. The value of $(25/8)\sqrt{T} A_i^{(K)}$ is stored in the location B(I,1), replacing the previous quantity set in KINT, which is no longer needed. The reciprocals of the elements B(I,L) are also computed and stored in BR(I,L).

The next section of subroutine KANDMU, from statement 20 down through statement 70, computes the quantities

$$A(1) = (25/8)\sqrt{T} \bar{a}^{(K)}$$

and

$$A(2) = (25/8)\sqrt{T} \bar{a}^{(\mu)}$$

(111)

from equation I(86), using the values of the matrix elements $a_{ij}^{(K)}$ and $a_{ij}^{(\mu)}$ which were previously stored in the lower halves ($I > J$) of the matrices Q(1,I,J) and Q(2,I,J)

by subroutine TRANSP. Since the values of $\bar{a}^{(\alpha)}$ for the viscosity and thermal conductivity are both computed from the same equation, the same coding can be used here for both properties, with the index $L = 1$ indicating values computed for the thermal conductivity and $L = 2$ values computed for the viscosity. In connection with this computation, it may be noted that the sum in equation I(86) need only be carried over values of $I > J$, since the matrix $a_{ij}^{(\alpha)}$ is symmetric and the diagonal terms do not contribute to the sum.

The final loop in subroutine KANDMU down through statement 90 computes the values of the subroutine arguments ZK(1) and ZK(2), which are proportional respectively to the translational thermal conductivity and the viscosity of the gas mixture. These values are computed from equation I(85), using the values of the elements B(I,L) and A(L) which were computed earlier in the subroutine

3.28 Subroutine KINT

The statements down through statement 10 of subroutine KINT compute the quantities

$$B(I,1) \equiv \frac{1}{k} \sqrt{T} \sum_{j=1}^n x_j \Delta_{ij}^{(1)} \quad (112)$$

and

$$\begin{aligned} B(I,2) &\equiv \frac{5}{6k} \sqrt{T} \sum_{j=1}^n x_j \Delta_{ij}^{(2)} \\ &= \frac{25}{8} \sqrt{T} ZM2(I) \sum_{j=1}^n x_j A_{ij}^{(\mu)} \end{aligned} \quad (113)$$

which are required in the transport property calculations. Because of the way in which the data $\Delta_{ij}^{(k)}$ are stored in the Q array by subroutine TRANSP, it is necessary to treat the terms with $i < j$ and with $i > j$ separately in these

computations; thus the statement preceding 10 carries out the sum over terms with $i > j$ and statement 10 the sum over terms with $i < j$. The diagonal terms with $i = j$ were added previously during the initialization of the sum by subroutine TRANSP.

Following the computation of the $B(I,1)$ and $B(I,2)$, subroutine KINT uses the computed values of the $B(I,1)$ in the small loop ending at statement 30 to calculate the value of the argument $ZKINT \equiv K_{int}/\sqrt{T}$ in units of milliwatts/cm- $^{\circ}K^{3/2}$, where K_{int} is the internal component of the thermal conductivity of the gas given by equation I(95).

3.29 Subroutine LIST

Subroutine LIST prints out certain portions of the input data for each case so as to provide a description of the problem which is to be solved by the code. Subroutine READ also provides some output of similar character.

Portions of the output normally produced by LIST are omitted if they would simply repeat the corresponding output for a previous case. If $SUPG\emptyset = .TRUE.$, the gas model is the same as in the preceding case, and the description of the gas model is skipped. If $SUPG\emptyset$ is $.FALSE.$, the following tables are printed:

- (1) The elemental composition of the gas in terms of "atom fractions", defined as the number of gram-atoms of each element per mole of the mixture of cold species. This table also lists the atomic weights assumed for the elements.
- (2) The data defining the forward rate constant for each of the reactions. This table also includes the criterion value CCHI for switching from the perturbation technique to numerical integration, and the third body matrix for the reactions involving third bodies.

- (3) A table which defines the chemical formula for each species by giving the $\alpha_{ij} = \text{LPIJ}(I,J)$ matrix. A separate column of this table contains 1 if a thermo fit is available for the species, 0 if not.
- (4) The matrix ν'_{ij} of the stoichiometric coefficients for the product side of all the reactions.
- (5) The matrix ν_{ij} of the stoichiometric coefficients for the reactant side of all the reactions.
- (6) A table listing the data for calculations of species thermal properties based on the physical model.
- (7) A table listing the electronic energy levels and degeneracies for all the species for which physical model data are provided.
- (8) A table of the thermo-fit coefficients for those species for which thermo fits are provided.

If an electronic nonequilibrium gas model is being used (NT=2), then a table giving the extra reaction data required by this model is inserted before the thermo-fit table (8).

After the gas model description has been printed out, LIST prints a summary of the input data for model and wedge calculations.

3.30 Subroutine MATINV

This is a matrix inversion subroutine. It is called by INGAS to compute the inverse A_{ki} of the matrix $\bar{\alpha}_{ij}$; see equation I(3) and associated discussion. MATINV computes the inverse of matrix A using the Gauss-Jordan reduction with the maximum pivot strategy (ref. 8).

3.31 Subroutine MØDEL

Subroutine MØDEL is the principal routine for calculations of conditions on models in the NATA code. MØDEL performs normal-shock calculations, computes conditions at the

inviscid stagnation point, calculates stagnation point heat fluxes, and provides output of these results. In addition, if conditions on wedge models are to be calculated, MØDEL calls subroutine WEDGE. The overall structure of MØDEL is diagrammed in figure 14.

3.31.1 General Description

Immediately upon entry, MØDEL performs some checks to determine whether model condition calculations should be done. Such calculations are skipped if both of the logical variables AXISYM and WEDGEM are false, or if the free-stream Mach number is less than 1.5. The latter restriction is imposed because the normal shock solution fails to converge when the Mach number is below some value in the range from 1.5 to about 2. Models are not normally tested at such low Mach numbers in arc-heated wind tunnels because the corresponding nozzle diameter is too small to accommodate models of reasonable size.

If model calculations are to be done, subroutine ELTIME is called with the argument IP = 0 to record the execution time at the beginning of the model calculations. ELTIME is called again, just prior to the RETURN, to determine the CPU time elapsed during the model calculations. This time is printed out. It is usually about 1 to 3 seconds.

Before beginning the normal shock solution, MØDEL sets a number of flow quantities which depend upon the free-stream flow conditions and are thus constant for a given call to MØDEL. When MØDEL is first called in a given flow solution, the equilibrium and frozen shock density ratios EPSLØN and EPSF have the value zero, set in the calling routine (FRØZEQ or NØNEQ). These ratios are reset to the value 0.01. When MØDEL is called with EPSLØN = 0, it computes the Prandtl number at the model surface temperature, PRW, based on the cold species mole fractions QPJ(K). This value of PRW thus assumes that the gas at the model surface is in a state of thermochemical equilibrium at the surface temperature.

Most of the calculations in MØDEL are contained within a large DØ loop with the DØ index ISØLN, beginning below

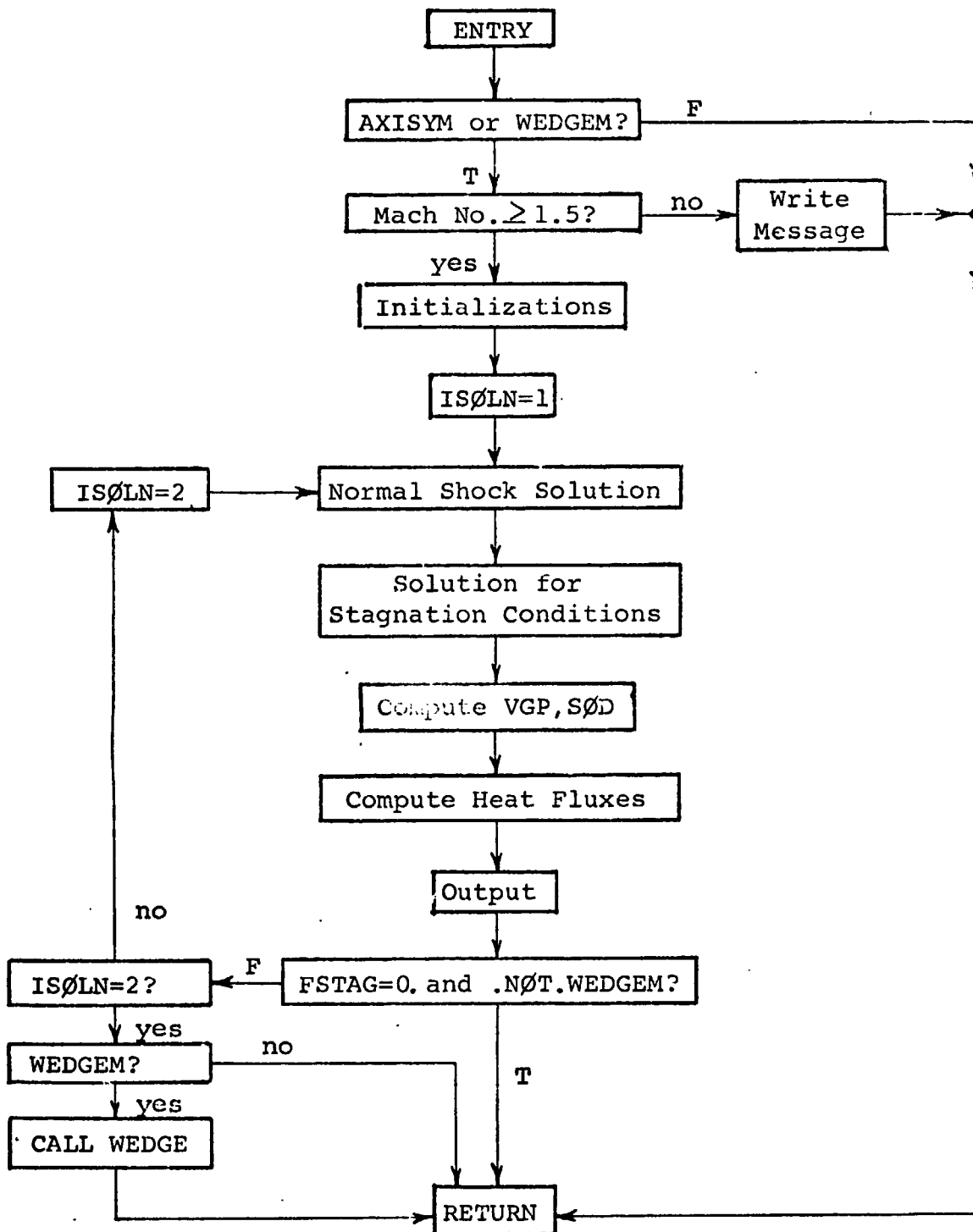


Figure 14. Overall Flowchart of Subroutine MDEL

statement 40 and ending at 440. For ISØLN = 1, the normal shock is calculated assuming chemical equilibrium behind the shock. For ISØLN = 2, the flow across the shock is assumed to be chemically frozen. If the control variable FSTAG is 0, the frozen shock calculations are skipped unless wedge calculations are to be done (WEDGEM = .TRUE.). If wedge calculations are called for, the frozen shock solution is carried out, regardless of the value of FSTAG, to provide the frozen stagnation temperature which is required for use in subroutine WEDGE. If FSTAG is negative, the equilibrium shock calculations are skipped.

For either value of ISØLN, the conditions behind the normal shock and those at the inviscid stagnation point are calculated using methods discussed below in Sections 3.31.2 and 3.31.3. Then the stagnation-point velocity gradient parameter $(R_p/u_1)(du_e/dx)_s$, denoted by VGP(L), is computed for spherical or cylindrical nosed models (L = 1) and for flat-faced models (L = 2). The ratio SOD(L) of the shock standoff distance to the model radius is also calculated for L = 1 and 2. These calculations are based upon equations I(463) to I(465) for the velocity gradient parameter, and I(475) to I(478) for the standoff distance.

Next, the various heat flux values discussed in Section 8.1.4 of Volume I are computed. In preparation for these calculations, the dissociation enthalpy h_p of the gas at the inviscid stagnation point (denoted by HCF) is computed by summing the product of the mole fraction and the enthalpy of formation SHJA(I) over all neutral atomic species (see equation I(544)). In the case of an equilibrium shock, the mole fractions SAVEC(I) at the stagnation point are obtained from the common array ZCAP(I), which is set in subroutine EQCALC. In the case of a frozen shock, the mole fractions are the same as those in the free stream flow ahead of the shock, and are calculated from the concentrations GJ(I) using equation I(1). In the evaluation of HCF, the sum is restricted to neutral atoms by rejecting all species containing more than one element, more than one atom of an element, or a negative number of atoms of an element.* This algorithm

*Positive ions are represented as compounds containing a negative number of electrons.

accepts the electron species, which of course is not a neutral atom, but this does not affect HCF because the formation enthalpy of the electron is zero.

The transport properties μ_e (VISC), N_{Pr} (PR), N_{Le} (LE), and σ (SIGMA) at the inviscid stagnation point are computed by calling subroutine TRANSP. The heat fluxes are calculated using equations I(466), I(468) and I(469) with suitable conversion factors to obtain English engineering units in the output. For example, the numerical coefficient in the Fortran formula corresponding to equation I(466) is obtained as follows:

$$0.51 = \frac{1.8 \times (30.48)^{3/2}}{453.6} \times 0.763$$

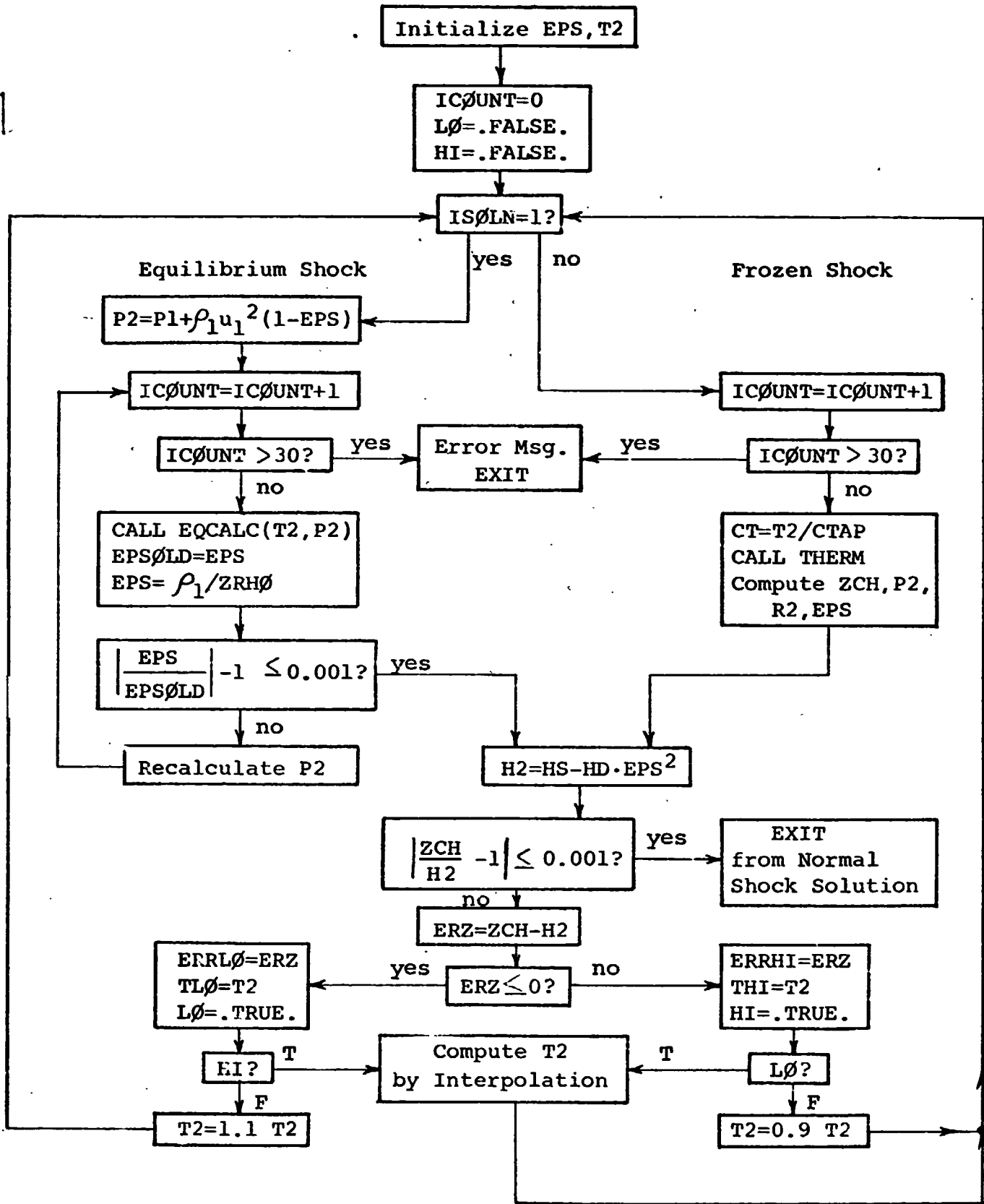
The factor of 1.8 converts the enthalpies from internal code units of cal/gm to Btu/lb; $(30.48)^{3/2}$ converts cm^2/cm^2 to ft^2/ft^2 ; and $(453.6)^{-1}$ converts g to lb.

After the heat fluxes have been computed, the results of the calculations are printed out. To this end, the results are first loaded into the array SCØUT, to allow use of a relatively compact output format. If a binary output tape is being prepared during the run (DATAPE = .TRUE.), certain of the model condition results are also written on tape unit ITPØUT.

3.31.2 Normal Shock Solution

Figure 15 diagrams the method used in the normal shock solution. This technique is explained in analytical terms in Section 8.1.1 of Volume I. The solution is based upon an iteration to determine the temperature T2 behind the shock. The value of T2 obtained during the previous call to MØDEL is used as a first estimate; this value is stored as T2E (equilibrium shock) or T2F (frozen shock) in common block /STAG/. Prior to the first call to MØDEL, these values are initialized to the reservoir temperature.

The criterion for selection of T2 is that the static enthalpy behind the shock, calculated from T2 using equation



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Figure 15. Flowchart of the Normal Shock Solution

I(429b), be equal to the value H2 obtained from the Rankine-Hugoniot energy equation for the shock, equation I(432b). In the case of a frozen shock, the gas is thermally perfect because the mole fractions are constant. The static enthalpy ZCH is calculated, in this case, from species enthalpies obtained by calling subroutine THERM after the nondimensional temperature CT has been set to the value T2/CTAP. The pressure P2 behind the shock is then computed from I(437), and the shock density ratio from I(434).

The equilibrium shock solution is more complicated because in this case the density ratio ϵ is not given as an explicit function of T2. Thus, a double iteration is required. For each trial value of T2, ϵ and the pressure P2 are calculated by an inner iteration as described in the discussion around equation I(438). In this case, the static enthalpy ZCH and density ZRH \emptyset behind the shock are obtained by calling subroutine EQCALC. The outer iteration to determine T2 is the same as for the case of a frozen shock.

The object of the outer iteration is to make ERZ = ZCH - H2 equal to zero, to within an accuracy of 0.1 percent. The temperature T2 satisfying this criterion is obtained by iterative linear interpolation:

$$(T2)_{\text{new}} = (T2)_{\text{old}} - ERZ \cdot \frac{THI - TL\emptyset}{ERRHI - ERRL\emptyset} \quad (114)$$

Here $(T2)_{\text{new}}$ is the improved estimate resulting from the linear interpolation, $(T2)_{\text{old}}$ is the previous estimate, ERZ is the enthalpy error obtained at $(T2)_{\text{old}}$, ERRHI is the last positive error ERZ obtained and THI the $(T2)_{\text{old}}$ corresponding T2 estimate, and ERRL \emptyset is the last negative error obtained with TL \emptyset the corresponding temperature. Of course, this formula cannot be used unless points with both positive and negative ERZ have been computed. This requirement is implemented with the aid of logical variables L \emptyset and HI which are initialized to .FALSE.. L \emptyset is set to .TRUE. when a negative ERZ is obtained, and HI is set to .TRUE. when a positive ERZ is found. If ERZ is negative and HI is "false", T2 is increased by 10 percent. If ERZ is positive and L \emptyset is "false", T2 is

decreased by 10 percent. The interpolation formula (114) is not used until both $L\phi$ and HI are .TRUE..

3.31.3 Conditions at the Inviscid Stagnation Point

After the normal shock calculation has converged, the conditions at the inviscid stagnation point are computed as explained in Section 8.1.1 of Volume I. A flowchart of the coding is given in figure 16. The logic is similar to that of the normal shock solution (figure 15), but with two simplifications:

- (1) The stagnation pressure is calculated from the incompressible Bernoulli equation (428b) with no iteration. Thus, only a single iterative loop is required even for the case of equilibrium flow.
- (2) The initial estimate of the stagnation temperature T_S is taken to be the temperature T_2 just behind the normal shock. Since T_2 is known to be lower than T_S , the logical variable $L\phi$ is not needed in the stagnation-condition solution. Linear interpolation is begun as soon as a positive ERZ is obtained, where ERZ is the difference between the known stagnation enthalpy H_S and the local enthalpy at the stagnation point (ZCH) computed from T_S and P_S .

3.32 Subroutine NEWRAP

The function of subroutine NEWRAP is to compute the flow conditions at a point in the equilibrium flow solution, using the method explained in Section 6.2 of Volume I. First, the system of $(c + 1)$ equations I(227) and I(252) is solved for the species mole fractions X_i and the pressure p . Then the remaining flow variables are computed from these data.

The Fortran notation in NEWRAP is similar to that in EQCALC, already discussed in Section 3.18, but there are some differences and additions. ISMCNR denotes the number of dependent species. Initially, this is equal to $n-c=ISS-ISC$.

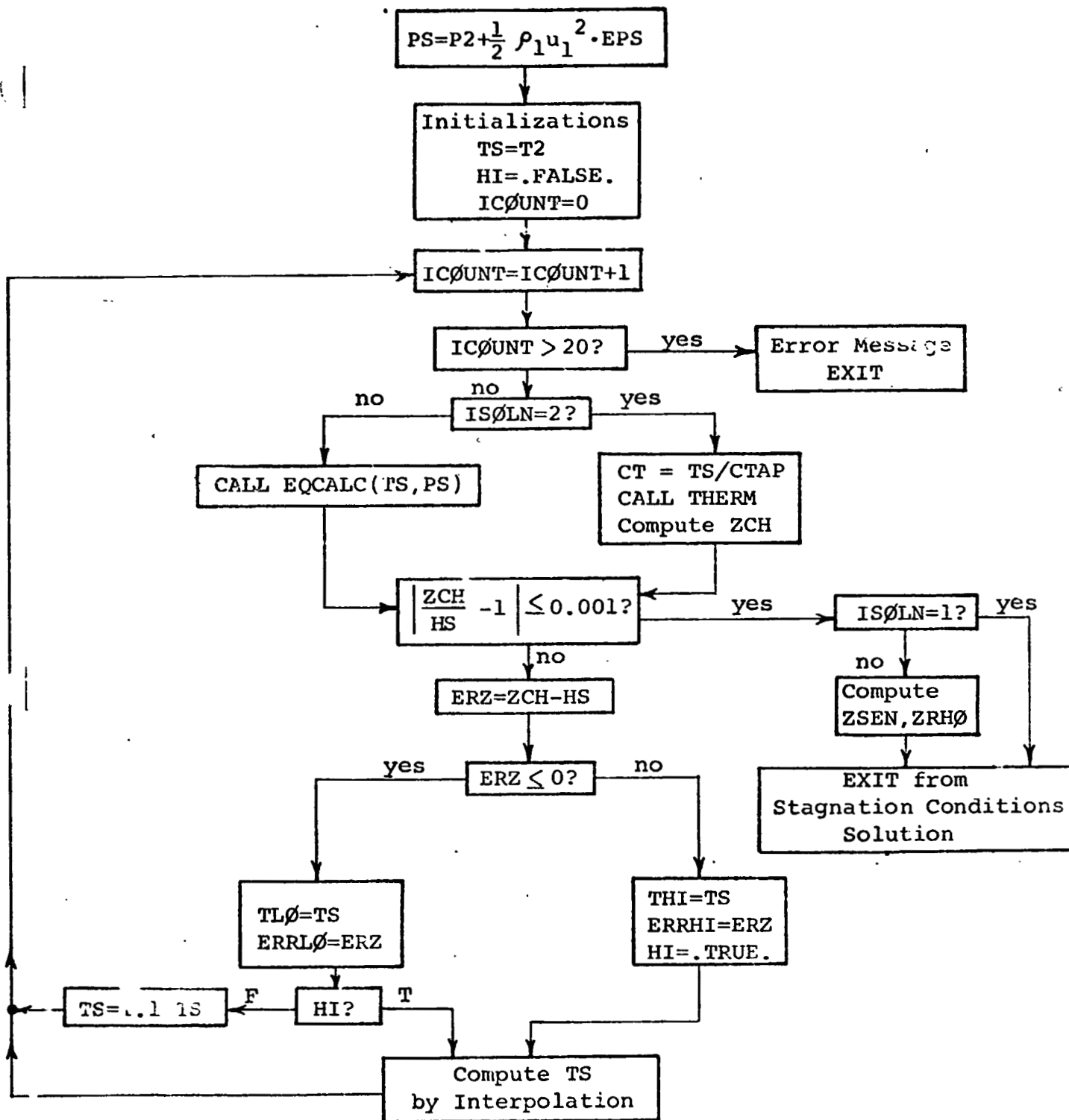


Figure 16. Flowchart of the Solution for Conditions at the Inviscid Stagnation Point

The array CHII is redefined as the logarithm of the equilibrium constants I(225) for the dependent species:

$$\text{CHII}(i-c) = - \frac{\mu_i^0}{R_0 T} + \sum_{j=i_m}^c \bar{v}_{i-c,j} \frac{\mu_j^0}{R_0 T} \quad (115)$$

where $i_m = \text{IM}$ is the index of the first independent species, initially equal to 1. A new array PGJ is defined by

$$\text{PGJ}(i) = \frac{S_i^0}{R_0} - \text{CRRB} \cdot W_i \quad (116)$$

where S_i^0 denotes the molar entropy of the i th species at the standard pressure, W_i is the molecular weight of the i th species, and*

$$\text{CRRB} = \frac{S_0}{R_0 W_0} \quad (117)$$

where S_0/R_0 is the nondimensional molar entropy of the gas in the reservoir and W_0 the reservoir molecular weight.

Immediately upon entry, NEWRAP calls subroutine THERM to compute the species thermal properties at the temperature $CT \cdot T_0$ corresponding to the current value of the nondimensional temperature CT . NEWRAP then sets up the arrays CHII and PGJ, defined above, and obtains a set of initial estimates of the species mole fractions $X_i = \text{CAPX}(I)$ from equation I(1):

$$\text{CAPX}(i) = W \gamma_i \quad (118)$$

The molecular weight $W = \text{CM}$ and species concentrations $\gamma_i = \text{GJ}(i)$ are the values in storage. For the first call to NEWRAP, these are reservoir values. In subsequent calls, they are the final values for the preceding flow point. Also,

*CRRB is set in subroutine INGAS following the entry INTA.

SKIL(I) is initialized to CAPX(I), and ZP to $\ln(\text{PRES})$, where PRES is the stored value of the nondimensional pressure.

The Newton-Raphson iterative loop starts at statement 70. The system of linear equations to be solved in each step of the iteration is I(258). The constant terms $F_j(Y_k^F)$ are loaded into the M2th column of the matrix AA, where $M2 = c + 2$. For $j = 1$ to c , these terms are given by equation I(227), and are computed in the DØ loops ending at statement 90. The constant term for $j = c + 1$ is given by I(252), and is loaded into AA(M1,M2) by the statements between 90 and 100. In the loop "DØ 100", the upper limit ISSNR is the current number of species included, initially equal to ISS.

The matrix of coefficients for the system of equations I(258) is loaded into the first $(c + 1)$ rows and columns of AA. The coefficients I(233b) for $j = i_m$ to c and $k = i_m$ to c go into the first c rows and columns, and are set by the nested DØ loops ending with statement 150. The coefficients I(261b) for $j=c + 1$ and $k = i_m$ to c are computed in the DØ loops ending at statement 170. The coefficient I(261c) for $j = c + 1$ and $k = c + 1$ is set up by the loop DØ 180. Finally, the coefficients I(261a) for $j = i_m$ to c and $k = c + 1$ are loaded into AA by the DØ loops ending at statement 210.

If the number of dependent species ISMCNR is zero a logical indicator NØDEPS is set to .TRUE. at the beginning of NEWRAP. Tests on NØDEPS then cause the subroutine to skip all of the DØ loops with DØ indices running from 1 to ISMCNR, in which sums from $c + 1$ to n are evaluated.

During an equilibrium expansion to high Mach numbers, the mole fractions of ionized species can become extremely low as the gas temperature falls to a small fraction of its value in the reservoir. To avoid computational problems arising when the electron mole fraction becomes exceedingly small, the electrons and all ionized species are eliminated from the equilibrium calculation whenever the electron mole fraction drops below 10^{-20} . The test for this condition is in statement 220. If CAPX(1) is found to be $\leq 10^{-20}$, the following

actions are taken:

- (1) The indicator JJK is reset from its initial value of 0 to 1. The latter value signifies that the elimination of charged species has already been performed.
- (2) The number of ions (IC) is subtracted from ISMCNR and ISSNR.
- (3) IM is increased from 1 to 2.
- (4) AA(1,1) is set to 1, and all the other entries in the first row and first column of AA are set to 0.
- (5) The electron mole fraction, CAPX(1), is set to 0.
- (6) The mole fractions of all the ion species are set to 0.

Regardless of whether the charged species have been eliminated, the system of equations I(258) is solved by calling subroutine DSMSØL. Improved values for the species mole fractions and the pressure are then obtained using equations I(262). The correction factor $1 + h_n^r$ in I(262) is denoted by ZB for $n = 1$ to c (i.e., for the mole fractions), and by ZC for $n = c + 1$ (the pressure). If any of these correction factors turns out to be 0 or negative, the corresponding quantity (mole fraction or pressure) is adjusted by dividing it by 2. Based on the improved values of p and the X_i , the other quantities in the calculation ($ZP = \ln p/p_0$ and $CGMU(K) = \ln X_k$), are then recomputed. Next, the mole fractions for the dependent species are computed from the equilibrium relations I(224), which are written in the logarithmic form

$$\ln X_i = \ln K_{pi} + (\nu_{i-c}^* - 1) \ln p + \sum_{\ell=1}^c \bar{\nu}_{i-c,\ell} X_\ell \quad (119)$$

The term $\ln K_{pi}$ is denoted in NEWRAP by CHII(i), equation (115). The $\ln X_i$ are computed in the array SKIL(i-c) and the resulting values are saved in CGMU(i). Then the X_i themselves are computed by taking exponentials and are stored in SKIL(i-c) and in CAPX(i).

The convergence criterion for the iteration is that the absolute magnitudes of all of the relative corrections h_n^r be smaller than or equal to TEST = 10^{-6} . When convergence has been achieved, the other flow variables are computed from the pressure, the mole fractions, and the known reservoir conditions. The relations used in these calculations are equations I(236), I(237), and I(242) for the molecular weight, density, and specific enthalpy, equations I(79b) and I(81) for the effects of gas imperfections, equation I(263) for the flow velocity, and equation I(264) for the mass flux. As noted in Section 3.22.1, the flow variables are computed in nondimensional form.

3.33 Subroutine NEXTMP

The function of subroutine NEXTMP is to determine the location (XMØDEL) of the next point at which model condition calculations are to be done. In the case of channel flow solutions, the routine determines the location corresponding to the next specified channel width.

The first two arguments of NEXTMP specify which of the model points called for in the input have already been passed in the current solution. ITS is the index of the next of the specified test-section diameters or channel widths, while XMØD1 is the x-coordinate the next model point in the geometric sequence of model points.

In the case of channel flow (NPRFLS = 2), the geometric sequence is not used. NEXTMP simply calls subroutine FINDXC to determine the position XMØD2 at which the width of the wide (MBLth) profile has the specified value TCAR(ITS), and sets XMØDEL equal to XMØD2.

In the case of nozzle flow (NPRFLS = 1), the next model point may be either the next point (XMØD1) in the geometric sequence or the point with the next specified area

ratio. In this case, NEXTMP calls FINDX to determine the location (XMØD2) of the point with the area ratio TSAR(ITS), and sets XMØDEL to the lower of the two values XMØD1, XMØD2. If the next point turns out to be at the specified area ratio, ITS is incremented by 1 to signify that TSAR(ITS) has been used. If the next point is a point in the geometric sequence, XMØD1 is replaced by FACMP times XMØD1, which is the next subsequent value of the sequence.

3.34 Subroutine NØNEQ

NØNEQ is the controlling routine for the nonequilibrium flow solution. The overall method of solution has been explained in Section 7.2 of Volume I. Briefly, the solution is started upstream of the throat, near the reservoir, using the perturbation method. At each step in the perturbation solution, the quantities $\xi \chi_i$ (which are measures of the departures of the reactions from equilibrium) are calculated and tested. When any one of these quantities reaches a specified size, the perturbation technique is abandoned and the solution is continued by numerical integration. If the integration is begun well downstream of the throat, it is carried out using the normal, direct method in which the specified nozzle geometry is taken to be a condition on the flow. However, if the integration is started upstream of the throat, an inverse method is used, in which the equilibrium density distribution $\rho(x)$ is used to define the flow. In this case, a switch is made from the inverse method to the direct method when the solution reaches a point slightly beyond the throat.

Figure 17 is an overall flowchart of subroutine NØNEQ. This flowchart shows how the various methods are organized to produce the entire solution. The operations enclosed in double boxes are represented by separate flowcharts and discussed below. IUPD is an indicator which is initialized to 1 (in subroutine READ), and is reset to 0 (in subroutine THRØAT) at the switch from the upstream to the downstream region. INEQ is another indicator, initialized to 0, and set to 1 when the numerical integration is started. Figure 17 does not show any exits from the control loops that generate the solution, because the case termination controls are within the boxes labelled "Numerical integration" and "Perturbation solution". The statements involving MØDLPT, XMSET, and subroutine NEXTMP pertain to the calculation of conditions at model points, and are discussed below.

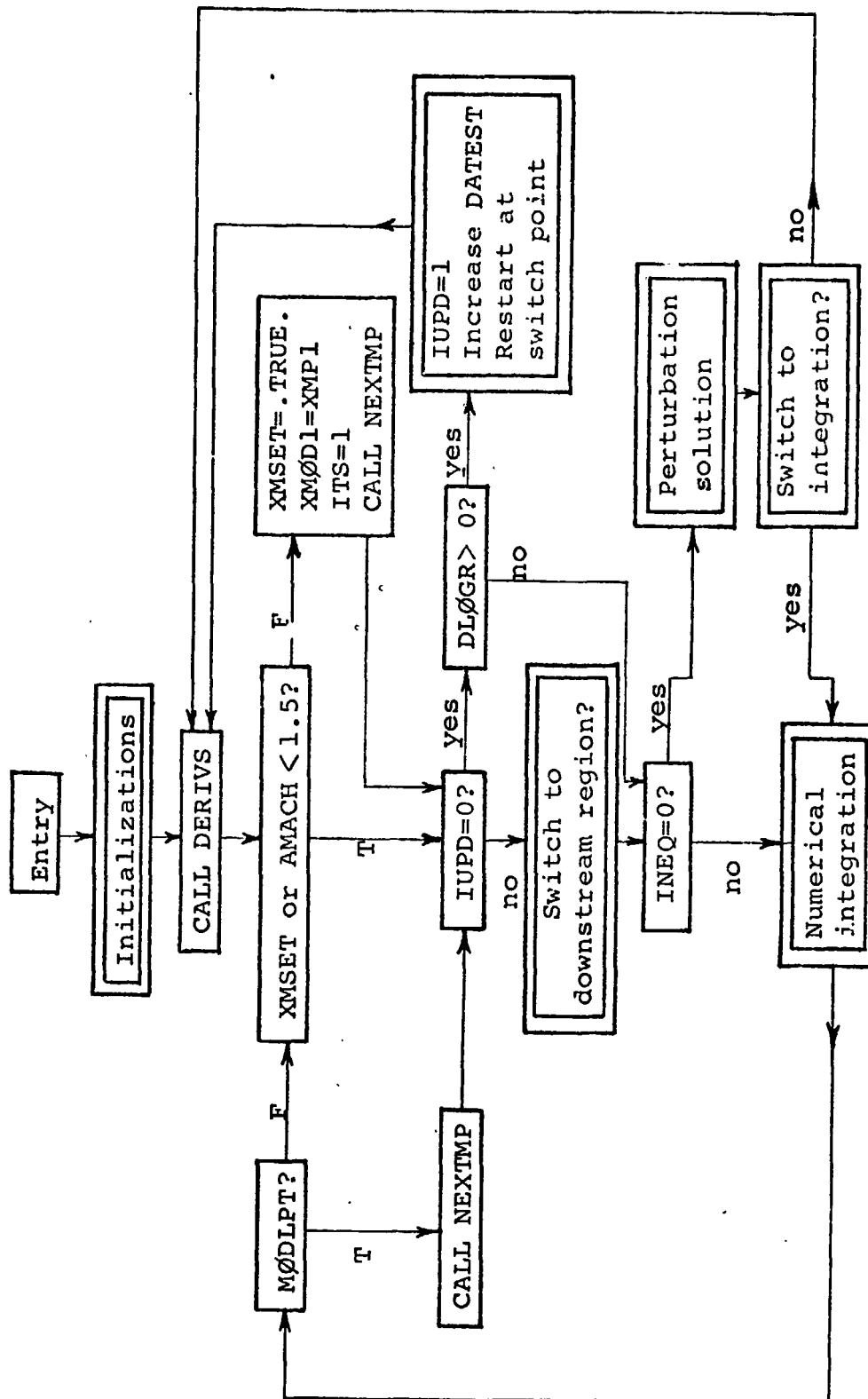


Figure 17. Flowchart of Subroutine NØNEQ

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The remaining parts of this section discuss individual portions and aspects of subroutine NØNEQ in detail.

3.34.1 Initializations

Figure 18 is a flowchart of the initializations section of NØNEQ, represented by the uppermost double box in figure 17. This flowchart shows only the initializations of control variables. In the subroutine, several physical variables are also set, but their values are not needed for following the logic.

The perturbation technique, which is used to start the solution, proceeds by taking steps in temperature, ΔT . The initial temperature decrement is given by $\Delta T/T_0 = \text{DELTL}$, where the variable DELTL is under input control and has a preset value of 0.01. This preset value of DELTL gives fairly large steps in the perturbation solution. In cases including the boundary layer ($\text{ISW3B} \neq 0$), to allow more accurate evaluation of the boundary layer integral $I(172)$, DELTL is reset to $0.049 (1 - \text{CTMAX})$, where CTMAX is the nondimensional sonic temperature T^*/T_0 based on the equilibrium solution. This value gives 20 steps upstream of the throat if the perturbation technique is used all the way to the throat. If the numerical integration is started upstream of the throat, as is usually the case, the number of steps is larger because the initial step size in the integration is much smaller than the initial perturbation step size.

The temperature, species concentrations, and gas properties are initialized for the nonequilibrium solution in the main program. Before calling NØNEQ, the main program sets the nondimensional temperature CT equal to $1. - \text{DELT}$, where DELT is set by the same rules used in calculating DELTL in NØNEQ. MAIN then calls NEWRAP to determine the equilibrium flow conditions at this temperature, calculates the effective area ratio, and determines the corresponding position in the nozzle by calling FINDX.

As explained in Section 7.3 of Volume I, the system of equations used in the perturbation technique is solvable only if the β_{ij} matrix has a rank of $n-c$. The initializations section of NØNEQ computes the rank IX of BETA(I,J) and executes an error exit if IX is not equal to $n-c$.

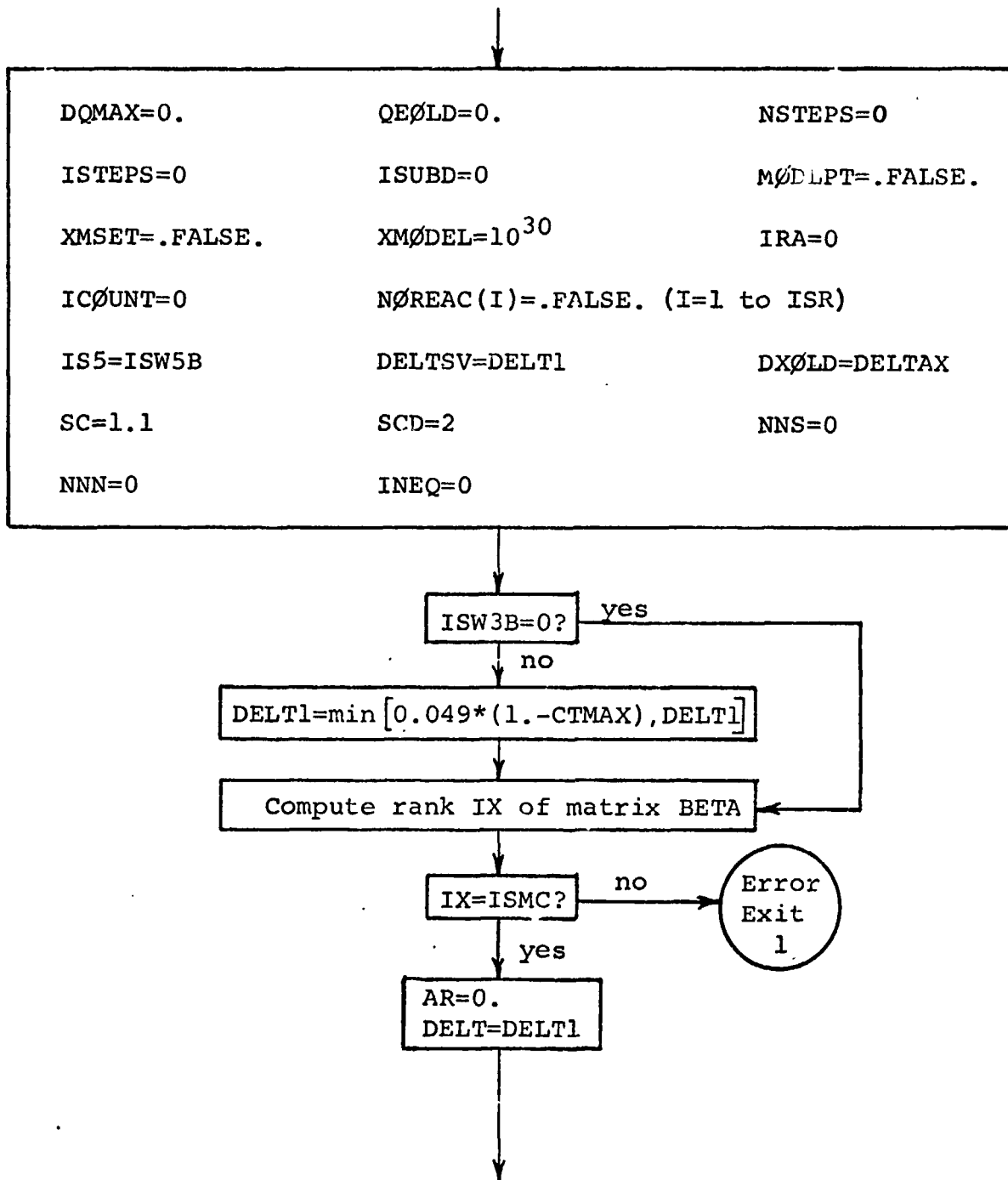


Figure 18. Flowchart of Initializations Section of Subroutine NØNEQ

3.34.2 Switch to Downstream Region

Figure 19 is a flowchart of the operations represented by the box labelled "Switch to downstream region?" in figure 17. The coding shown in figure 19 determines whether the conditions for switching to the downstream solution have been satisfied, and if so, carries out the switch by calling subroutine THRØAT. The conditions are as follows:

- (1) $DLØGA = d \ln \tilde{A} / dx$, where \tilde{A} is the area ratio computed using the inverse method, must be nonnegative.
- (2) The position of the flow point must lie beyond the geometric throat ($CX > 0$).
- (3) The area ratio $\tilde{A} = AFNX$ must be greater than or equal to a value DATEST which is preset (in subroutine INIT) to 1.01.

When all of these conditions have been satisfied, a large number of flow variable and control parameters are saved in the arrays TB, ITB, and BLBK, and the switch to the downstream region is executed by calling THRØAT. In THRØAT, IUPD is reset to 0 and RSA is set to a value which rescales the actual nozzle geometry to make it consistent with \tilde{A} .

In cases including the boundary layer, the coding in figure 19 also sets the ØMDST(I) array based on the displacement thickness DELBL(I) at the first point where the conditions (1) and (2) are satisfied.

3.34.3 Restart of the Upstream Solution

Figure 20 is a flowchart of the operations represented by the box labelled "IUPD = 1, Increase DATEST, Restart at Switch Point" in figure 17. This section of NØNEQ is entered when a positive value of $d \ln \rho / dx = DLØGR$ is encountered in the downstream solution. If this condition arises so far beyond the upstream/downstream switch point that the effective area ratio $A_0 = AFNX$ exceeds DATEST by 5 percent of DATEST or more, an error exit is executed. Otherwise,

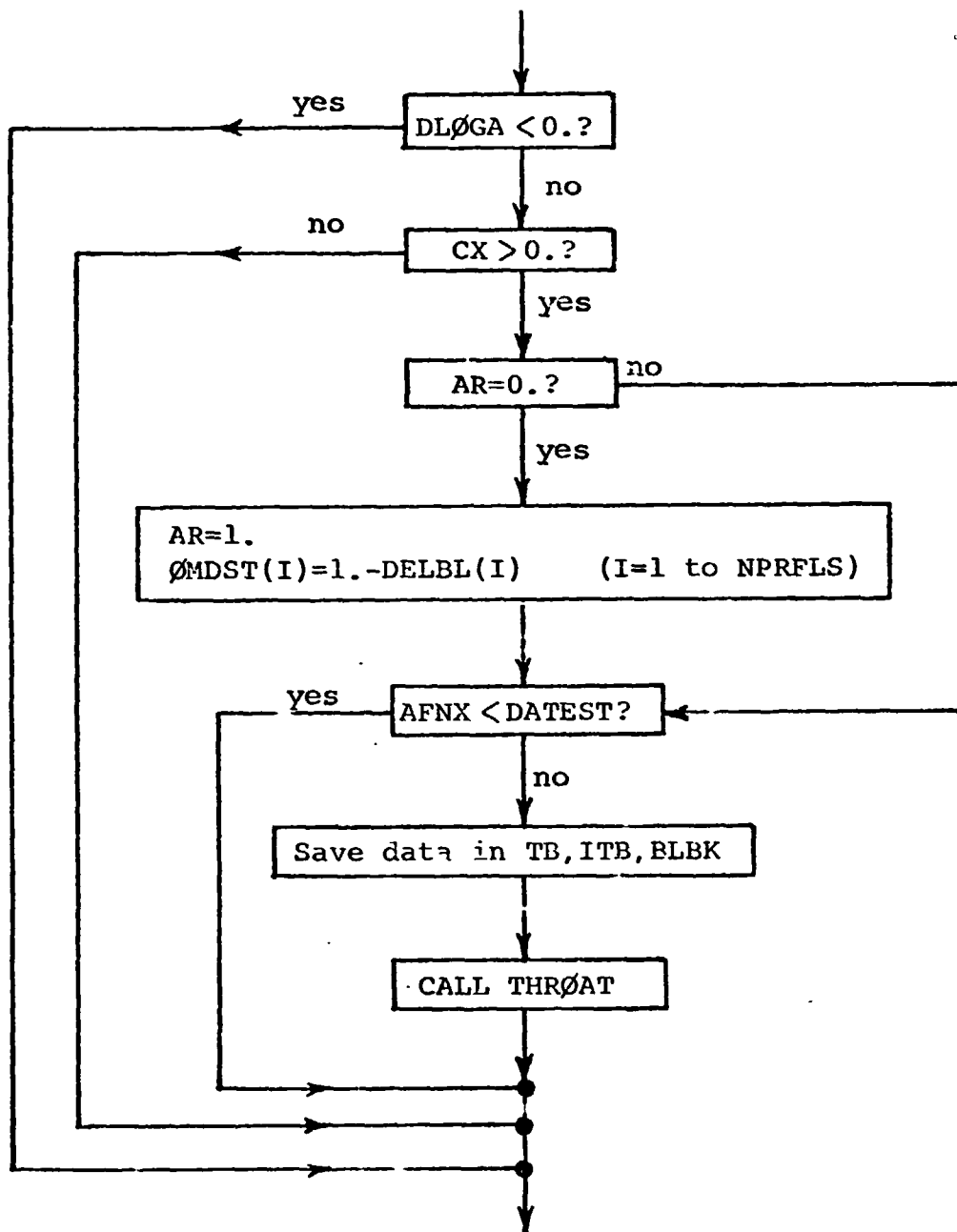


Figure 19. Flowchart of Switch to Downstream Region in Subroutine NØNEQ

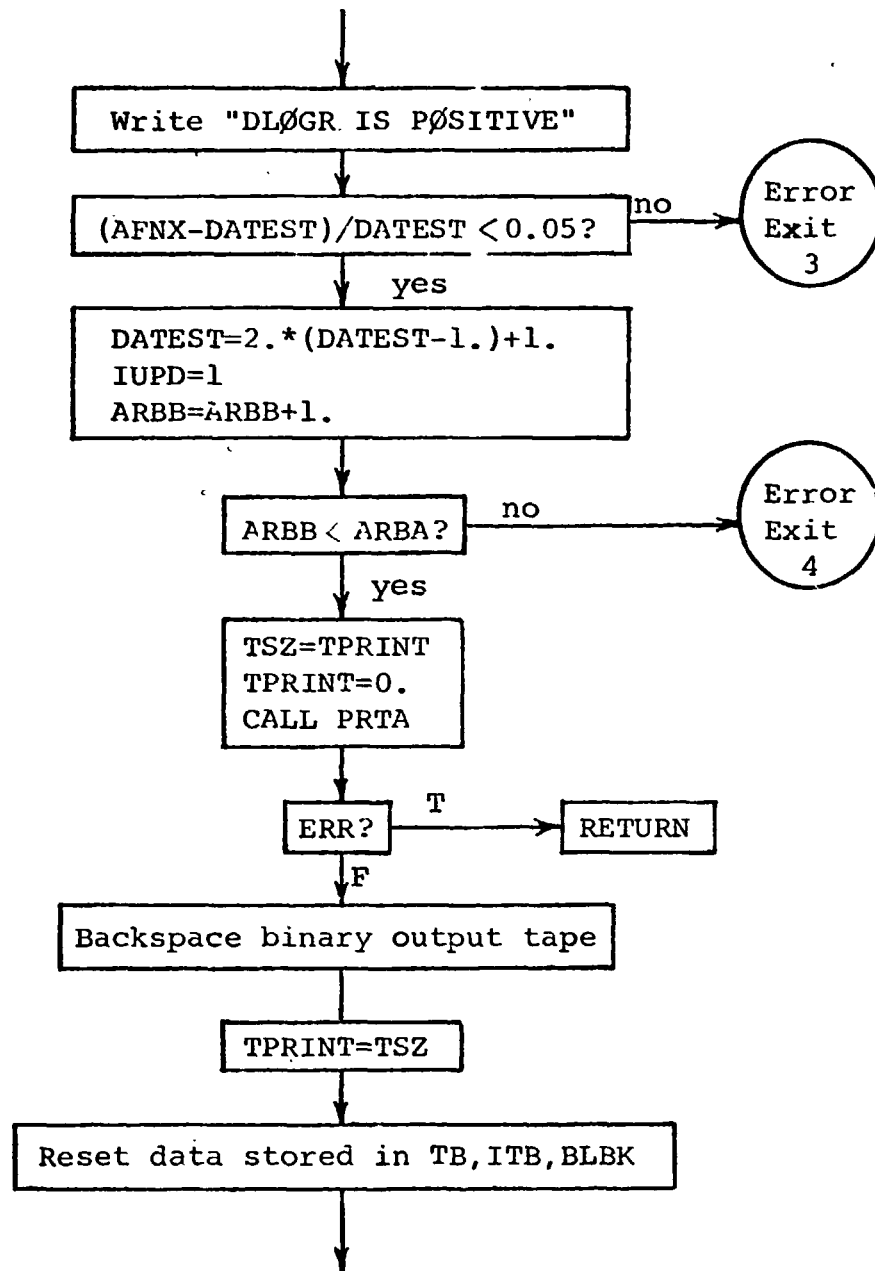


Figure 20. Flowchart of Restart of the Upstream Solution at the Switch Point in Subroutine NØNEQ

the program assumes that the positive $d \ln \rho / dx$ value is a result of switching to the downstream solution too near the throat, and attempts to recover the correct solution. To this end, the excess of DATEST over unity is doubled, IUPD is reset to 1 (signifying the upstream solution), the flow variables and control parameters are all reset to their values at the previous switch point, and the solution is restarted at that point. A counter ARBB (initialized to 0. in subroutine INIT) is also incremented by 1. If the $d \ln \rho / dx > 0$ condition is encountered repeatedly after successive restarts, the effort to continue the solution is abandoned after four attempts ($ARBB \geq ARBA = 5.$, set in INIT). Also, before the restart is executed, the conditions at the final point of the invalid downstream solution are printed out by setting TPRINT = 0. and calling PRTA. If a binary output tape is being produced for subsequent use by the NATA plot program, the records containing data from the invalid solution beyond the switch point are eliminated by backspacing the tape.

3.34.4 Perturbation Solution and Switch to Integration

Figure 21 is a flowchart of the operations represented by the boxes labelled "Perturbation solution" and "Switch to integration?" in figure 17. These operations may be summarized as follows:

- (1) Subroutine PERT is called to compute the perturbations PCT (in the nondimensional temperature), PERTGJ(J) (in the species concentrations), and SDCHI(I) (in the reaction parameters χ_i).
- (2) If the solution is in the downstream region (IUPD = 0) and the nondimensional temperature decrement DELT is still equal to the reduced value of DELT1 (see part 3.34.1 above), DELT is reset to the input value of DELT1, which was saved in DELTSV. This is done because the step size in the perturbation solution is sufficiently small in the throat and near downstream region, even when the larger decrement is used.

- (3) The largest $SDCHI(I)$ value ($DCHMAX$) and the smallest ($DCHMIN$) are determined.
- (4) The ratio $RATIO = DCHMIN/DCHMAX$ is examined. If this ratio is larger than or equal to a value $DCHRAT$ (preset in $BLØCK DATA$ to 10^{-4}), $NØNEQ$ proceeds to the checks for the switch to numerical integration, item (5) below. If $RATIO$ is less than $DCHRAT$, the reaction system assumed in the gas model will lead to a switch to integration at a point where some of the reactions are still very near equilibrium. The value of $RATIO$ is approximately independent of the size of $DCHMAX$; thus, the program responds to an excessively small value of $RATIO$ immediately, even though $DCHMAX$ may be much smaller or much larger than the value $CCHI$ at which the switch to integration is desired. To prevent the premature start of the numerical integration, $NATA$ artificially increases the rate constant for the reaction ($I = IMAX$) for which $|SDCHI(I)| = DCHMAX$, if this can be done without significantly affecting the solution. This procedure is discussed near the end of Section 7.3 in Volume I. If $RATIO < DCHRAT$, $NØNEQ$ determines whether this reaction involves a minor species (concentration less than or equal to $GAMIN$, preset to 10^{-10}) whose concentration is being decreased by the reaction. If so, the rate constant for the $IMAX^{th}$ reaction is increased by a factor of $RATIO2 = 1.1 * DCHRAT / RATIO$, and the perturbation calculation for the previously assumed temperature is repeated. If not, the perturbation solution is continued without adjusting the rate constant.
- (5) The next step is to check whether $DCHMAX$ has reached the value $CCHI$ at which the switch to the numerical integration is desired. If $DCHMAX$ is less than $CCHI$, the perturbation solution continues. If the axial coordinate

CX is greater than or equal to the current value of XMØDEL (the coordinate of the next model point), the indicator MØDLPT is set to .TRUE., and subroutine NEXTMP is called to determine a new value for XMØDEL. In any event, subroutine PRTA is called to provide output of the computed flow conditions at the current point of the perturbation solution. If MØDLPT is .TRUE., PRTA also calls subroutine MØDEL to calculate test conditions on a model.

- (6) After the return from PRTA, the case termination tests are applied. If CX is greater than or equal to CXMAX, or if the nondimensional temperature CT is less than or equal to TSTØP, control is transferred to statement 660. The flow conditions and model conditions at the final point are then printed and the RETURN is executed.
- (7) If case termination has not been reached, the temperature CT is decremented by DELT, and subroutine NEWRAP is called to compute the equilibrium flow conditions at the new temperature. The equilibrium effective area ratio AFNTS is then calculated from the continuity equation I(265), and subroutine AXFIT is called to determine the axial coordinate CX at this area ratio. Control is then transferred to statement 220 ("CALL DERIVS" in figure 17).
- (8) Once DCHMAX reaches or exceeds the criterion value CCHI for the switch to integration in step (5), NØNEQ commences an iterative adjustment of the nondimensional temperature CT in order to obtain a flow point at which $DCHMAX = |\delta \chi_i|_{\max}$ satisfies both of the conditions in equation I(381). Usually the perturbation solution overshoots the second condition I(381), i.e., DCHMAX is greater than PCTEST*CCHI, where

PCTEST = 1.2 (set in subroutine INIT). When this occurs, NØNEQ cuts DELT in half, adds DELT to CT in order to obtain a temperature halfway between the current value and the preceding one, and computes the corresponding point in the perturbation flow solution by calling NEWRAP and AXFIT as in step (7) above. This procedure, together with the normal procedure for advancing the perturbation solution (step (7)), repeatedly subdivides the temperature interval known to contain the flow points satisfying the conditions I(381).

- (9) When a point satisfying both of the conditions I(381) has been obtained, the switch from the perturbation technique to numerical integration is carried out. The indicator INEQ is reset from 0 to 1 and the perturbations PCT, PERTGJ(J) and PRHØ are added to the corresponding equilibrium flow quantities CT, GJ(J), and RHØ. The integration step size DELTAX is adjusted, if necessary, to make it no larger than $DXM = 0.01 * DCHMIN/DCHLL$, where the input DCHLL is preset to 10^{-4} in BLOCK DATA. This adjustment provides a reduced initial step size when $DCHMIN = \min_i |\Delta \chi_i|$ at the switch point is smaller than 10^{-4} , i.e., when some of the reactions are still very close to equilibrium. If this reduction in DELTAX were omitted, the normal step size controls of the integration would still bring DELTAX down to similar values at the cost of some additional computation. Any rate constants which have been artificially increased in step (4) are now restored to their correct values. Subroutine DERIVS is called to compute the derivatives and supplementary flow variables at the switch point, based on the perturbed temperature and concentrations, and control is transferred to the numerical integration section of NØNEQ. If the indicator FAILED is .TRUE. after the call to DERIVS, an error exit is executed. This indicator is .TRUE. if the flow conditions fail to pass certain validity checks in DERIVS and CØMM.

3.34.5 Numerical Integration

Figure 22 is a flowchart of the operations represented by the box labelled "Numerical integration" in figure 17. These operations may be summarized as follows:

- (1) The case-termination checks are the same as in the perturbation solution. If $CX \geq CXMAX$ or $CT \leq TSTOP$, control is transferred to statement 660. The flow conditions and model conditions at the final flow point are then printed, and the RETURN is executed.
- (2) If the end of the case has not been reached, the counter NNN is incremented and subroutine PRTA is called to provide output of the conditions at the flow point reached during the preceding integration step.
- (3) Next, the integration step size DELTAX and the factors SC and SCD used in changing DELTAX are adjusted. If NNN is equal to NQS (an input parameter, preset to 4), then DELTAX is multiplied by SC, NNS is incremented by 1, and NNN is reset to 0. Then if NNS is equal to NQS, SC is increased by 0.1, SCD is reset to the new SC value, and NNS is reset to zero. For $NNN < NQS$, no changes are made in these variables.
- (4) In preparation for the next integration step, the current values for CX, CT, GJ(I), and other quantities are saved in the storage locations CXB, CTB, GJB(I), etc. If the forthcoming step should fail for any reason, these data will be used in restarting the solution at the beginning of the step with a smaller step size.
- (5) Next, the anticipated position, $CX + DELTAX$, at the end of the forthcoming step is compared with $XMODEL$ and $CXMAX$. If $CX + DELTAX$ is greater than either of these values, DELTAX is reduced to make the end of the step coincide with the lesser of

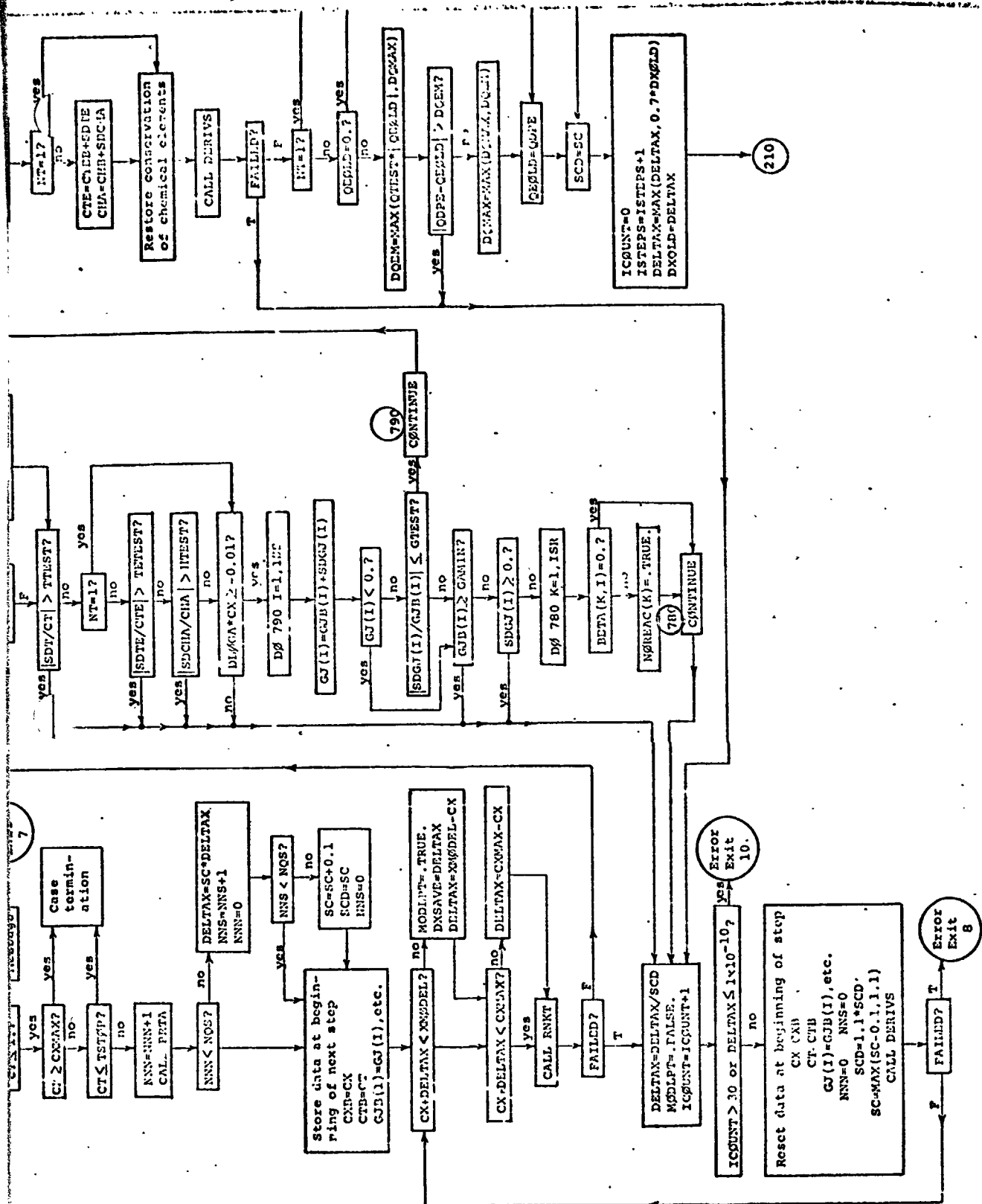


Figure 22. Flowchart of Controls for the Numerical Integration in Subroutine NNT0

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the two. If the end of the step will be at XMØDEL, then MØDLPT is set to .TRUE. and the DELTAX value before the reduction is saved in DXSAVE.

- (6) Subroutine RNKT is called to compute the changes in the dependent variables over the integration step.
- (7) The indicator FAILED is tested. A .TRUE. value of FAILED indicates that a validity check was not satisfied somewhere in RNKT, DERIVS, or CØMM. If FAILED = .TRUE., the step size is reduced by dividing DELTAX by SCD and the calculation is restarted at the beginning of the step, using the data saved in step (4). In the event of such a step restart, a counter ICØUNT is incremented. Also, MØDLPT is set to .FALSE., SCD is increased by a factor of 1.1, and SC is reduced by 0.1 (but not below a value of 1.1). In addition, NNN and NNS are both set to zero. The idea underlying these changes in the step control parameters is that the failure was caused by instability resulting from the use of an excessively large step size. Either the step has been increased too rapidly, or the solution is entering a region where the reaction system requires a smaller step size for stability. Thus, the step size itself is reduced, and the growth in step size is both temporarily arrested and reduced in rate. The increase in SCD allows the program to reduce the step size very rapidly if repeated failures are encountered in the same step, as sometimes happens in runs using electronic nonequilibrium models.
- (8) If FAILED is .FALSE. after the call to RNKT, additional validity checks I(407) and I(408) are applied to the results of the integration step. If any of these conditions is violated, the step is restarted with a smaller step size as described above in item (7). Also an indicator IFAIL is set to a distinctive value for each possible type

of failure. IFAIL is also set when step failures occur in other subroutines. Each time a step fails and is restarted, the IFAIL value is saved in the integer array IF(ICOUNT). These values can be obtained in the printed output by setting ISW5B to a negative value in the code input. Output of these data shows what tests, species, etc., are controlling the integration step size.* If a failure occurs in one of the conditions I(407a), I(407c) on the species concentrations, then the program checks whether the procedure of freezing the concentration of a minor species (Section 7.5.3 of Volume I) is applicable. The conditions for use of this procedure are that the species I, whose concentration GJ(I) violates I(407a) or I(407c), should have a concentration GJB(I) at the start of the step less than GAMIN, and that the concentration change for this species over the step, SDGJ(I), be negative. If these conditions are satisfied, all of the reactions contributing to changes in the concentration of this species are frozen by setting the corresponding elements of the array NPREAC(K) to .TRUE.. The step is then restarted in the usual way.

- (9) If the conditions I(407) and I(408a,b) are satisfied, the integration step is presumed to be valid subject to the condition I(408c), which is applied later. The changes SDT, SDTE, SDCHA in the dependent variables during the step are added to the values CTB, CTEB, CHB at the start of the step to obtain the corresponding values at the end of the step. The species concentrations are then adjusted to restore the original elemental composition of the gas mixture, as described in Section 7.1.3 of Volume I and in Section 3.34.6 below. Subroutine DERIVS is then called to compute the derivatives and supplementary flow variables at the end of the step.

*The numerical code used in IF(ICOUNT) is defined in Section 4.55.

- (10) If $NT = 2$ (electronic nonequilibrium model), the condition I(408c) on the change in the energy transfer to the electron gas (QDPE or \dot{q}_e) is now applied to the final QDPE value resulting from this call to DERIVS. If this condition is not satisfied, the step is restarted with a reduced step size in the usual way.
- (11) Once all of the validity checks have been passed, SCD is reset equal to SC and the step-restart counter ICOUNT is reinitialized to 0. The counter ISTEPS for successful steps is incremented by 1. The step size DELTAX is set equal to the larger of DELTAX itself and $0.7 \cdot DXOLD$. Thus, if DELTAX has been reduced sharply by repeated step restarts, it is here returned to 70 percent of its value in the preceding step.
- (12) Control is now transferred to statement 210 ("MØDLPT?" in figure 17).

3.34.6 Element Conservation

The adjustment of species concentrations to restore conservation of the chemical elements, explained in Section 7.1.3 of Volume I, is implemented by the coding from statement 800 down to 940 in NØNEQ. The programming is straightforward, and can easily be related to the formulas in Section 7.1.3 with the aid of the following summary of notations:

CJ	c_j'
DCI(J)	δc_j
GSQ(I)	$(\gamma_i')^2$
AIN(J,K)	A_{jk}
DCA(K)	$\frac{1}{(\gamma_k')^2} \sum_{j=1}^C \delta c_j \cdot A_{jk}$
AA(I,ISMCP1)	b_i (before call to DSMSØL)

CDIJ(I,K)	$\bar{v}_{i-c,k}$
AA(I,L)	a_{il}
AA(I,ISMCP1)	$\delta \gamma_{c+1}$ (after call to DSMSØL)

The system of equations I(346) is solved for $\delta \gamma_m$ by calling subroutine DSMSØL. After the adjusted concentrations of the dependent species ($i = c + 1$ to n) have been calculated from I(340) and those of the independent species ($i = 1$ to c) from I(338), the condition I(407a) is applied to the adjusted concentrations. If any of these concentrations is found to be negative, the step is restarted with a reduced step size in the usual way.

3.34.7 Model Points

The position of the next model point, at any point in the flow, is denoted by XMØDEL. Whenever a particular model point has been reached and the model calculations have been done, the new XMØDEL value is obtained by calling subroutine NEXTMP(ITS, XMØD1, XMØDEL). The argument ITS is the index of the next specified test section diameter, and XMØD1 is the coordinate of the next model point in the geometric sequence of model points. Subroutine NEXTMP, each time it is called, determines whether the next model point is at the test section diameter with index ITS or at XMØD1, sets XMØDEL accordingly, and updates ITS or XMØD1 as required. Before NEXTMP is called for the first time in NØNEQ, the initial values ITS = 1 and XMØD1 = XMP1 must be set. This is done by coding shown in figure 17, at the first point where the Mach number AMACH has reached or exceeded a value of 1.5. A .TRUE. value of the indicator XMSET denotes that this initialization has already been performed.

The indicator MØDLPT, if .TRUE., denotes that the current flow point in the perturbation solution or the integration is a model point. If MØDLPT = .TRUE., FPTA calls entry ØUT2 of subroutine ØUT1 to print the flow conditions (even though the TPRINT criterion may not be satisfied) and calls MØDEL to compute and print the model test conditions.

When a model point is encountered during the numerical integration, DELTAX is reset to provide a flow point exactly at the model point. During the perturbation solution, MØDEL is called at the first solution point beyond XMØDEL. To obtain a perturbation solution point coinciding with XMØDEL would require an iteration, as in the equilibrium and frozen flow solutions, because the flow is generated by decrementing the temperature, not by incrementing the position coordinate. Such an iteration has not been provided in the coding because, in the normal applications of NATA, the switch from the perturbation technique to the numerical integration always occurs upstream of the point where the Mach number is equal to 1.5.

3.34.8 Error Exits

Subroutine NØNEQ contains ten error exits, marked conspicuously with comment cards in the listing. Whenever one of these exits is executed, a message giving the error exit number is written out before the DUMP routine is called to provide diagnostic data. This message facilitates determining the nature of the error that caused NØNEQ to fail.

3.35 Subroutine NRMAX

This subroutine determines the sonic flow conditions for the case of equilibrium flow. Its logic and structure are practically identical with those for the corresponding calculation for frozen flow, performed in subroutine FRØZEQ. The flowchart for the frozen calculation (figure 9) is applicable to subroutine NRMAX. The discussion in Section 3.22.2 is also applicable. In the present case, the mass flux at each temperature is determined by calling subroutine NEWRAP.

3.36 Subroutine ØUT

This routine simply prints a list of definitions of the labels used in the output of the flow conditions and test conditions on models and wedges.

3.37 Subroutine ØUT1

Subroutine ØUT1 is the principal output routine of NATA. When called through its first entry point, it produces a printed list of the conditions at a special point in the flow, such as the reservoir or the sonic point. When called through its entry ØUT2(ISØLN), it prints a more comprehensive specification of the conditions at a general flow point. In addition, if the indicator DATAPE is .TRUE., a call to ØUT2 causes a record to be written on the binary output tape (tape 8), for subsequent use by the NATA plot program. Figure 23 is an overall flowchart of the subroutine.

3.37.1 Entry ØUT1

ØUT1 is called (by MAIN) to print the conditions in the reservoir and at the frozen and equilibrium sonic points. Since a call to ØUT1 precedes each of the three flow solutions, ØUT1 is also used to initialize some quantities used by ØUT2.

The chemically frozen specific heat ratio GAMMA is computed in a section of coding which is used by both ØUT1 and ØUT2. First, the entry THERM1 of subroutine THERM is called to compute the nondimensional species specific heats $C_{pi}/R_0 = CCPJ(I)$ at the current nondimensional temperature $CT = T/T_0$. The nondimensional molar specific heat of the gas mixture is then computed as

$$C_P \equiv \frac{C_p}{R_0} = \sum_{i=1}^n X_i \frac{C_{pi}}{R_0} \quad (120)$$

where the species mole fractions X_i are obtained from the array SAVEC(I), where they have been placed by the calling routine. Then the thermodynamic relation

$$C_p - C_v = R_0 \quad (121)$$

is used to obtain the specific heat ratio, γ :

$$\gamma = \frac{C_p}{C_v} = \frac{1}{1 - \frac{R_0}{C_p}} \quad (122)$$

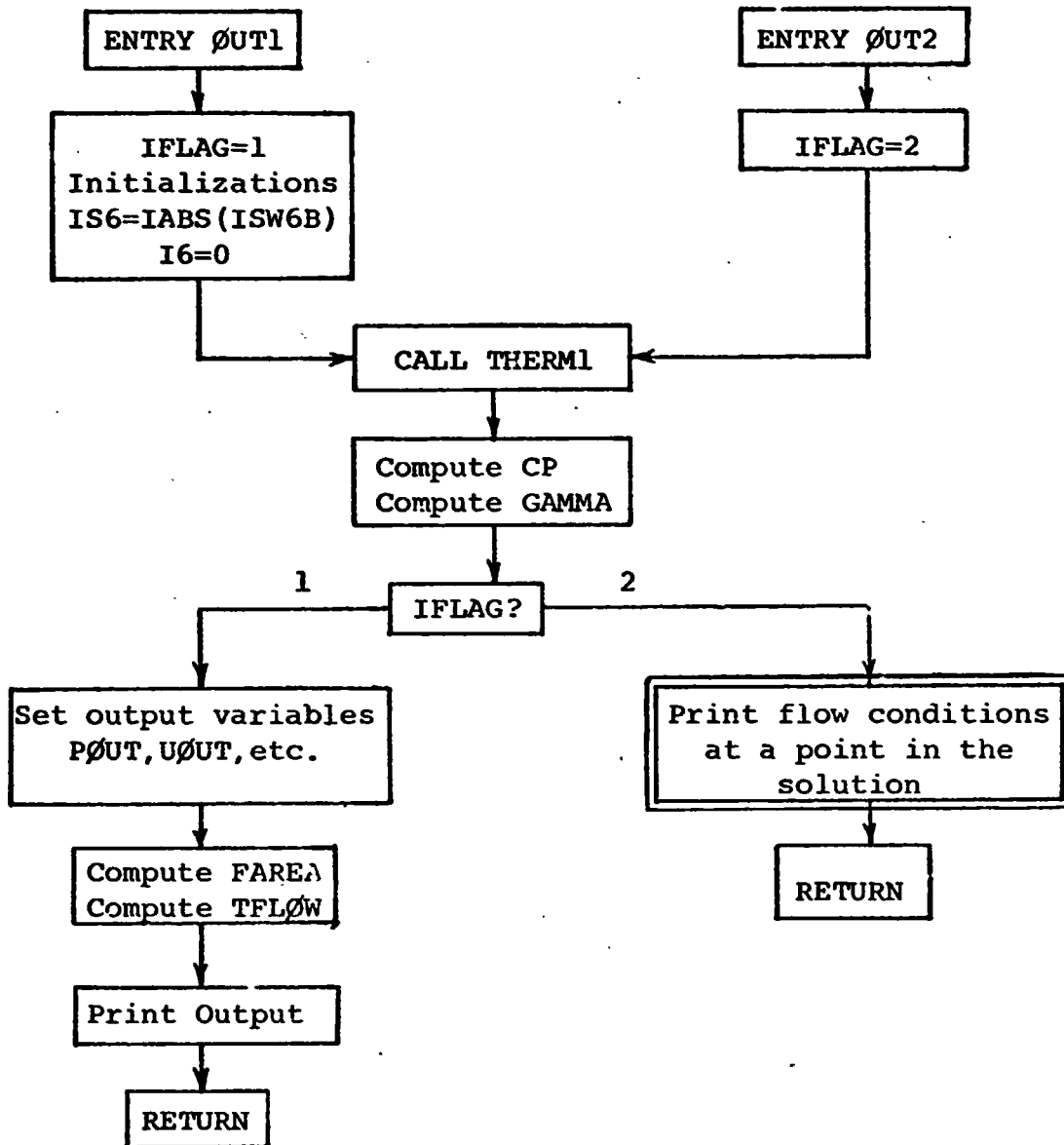


Figure 23. Overall Flowchart of Subroutine ØUT1

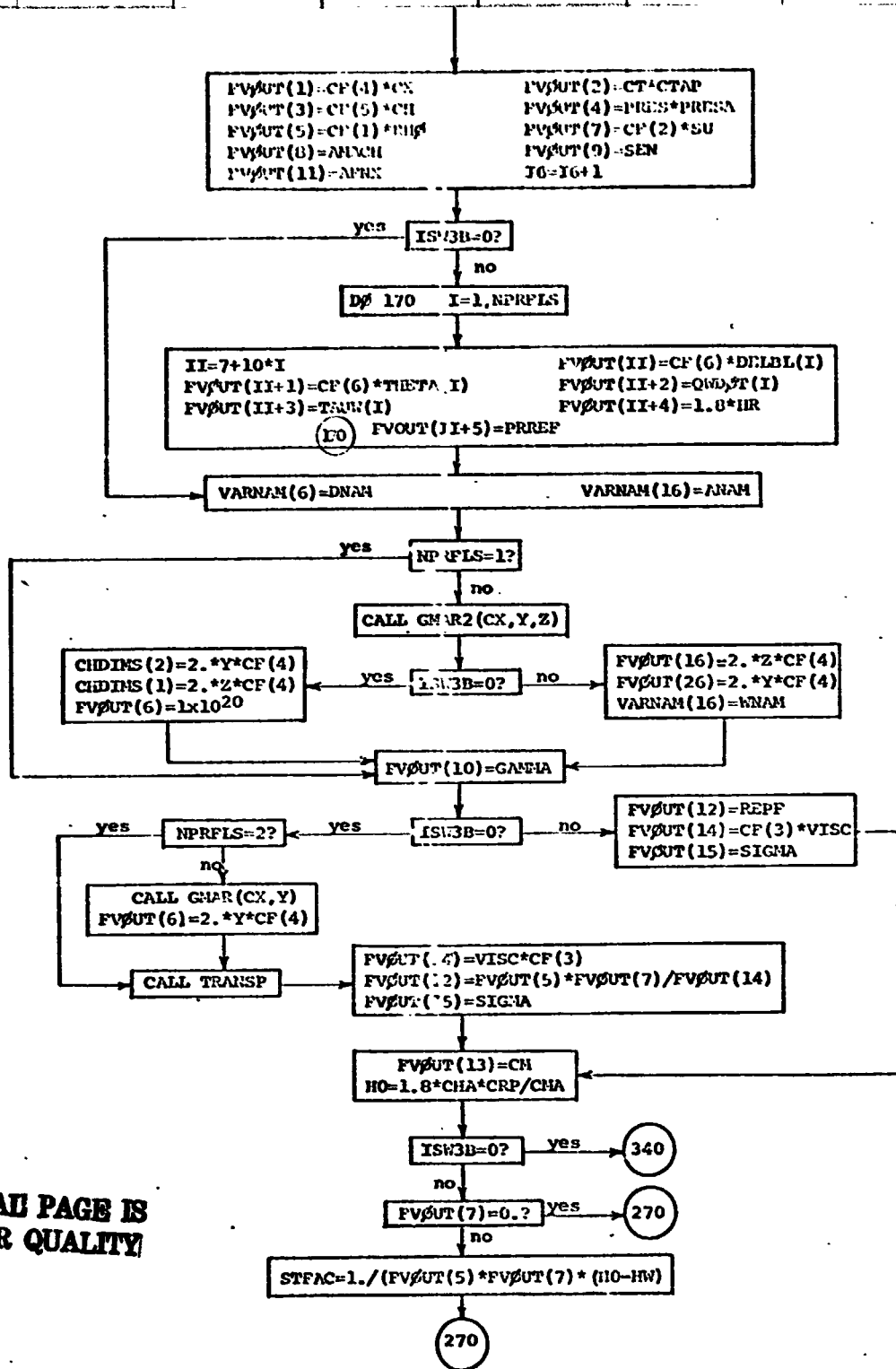
At the conclusion of this calculation, the indicator IFLAG is used to transfer control to the appropriate portion of the subroutine, depending on which entry point was called. If entry was through ØUT1, control passes to statement 30. The variables used for output of the flow conditions in ØUT1, namely PØUT, UØUT, etc., are then set. Next, the effective cross sectional area of the flow at the throat, FAREA, is computed. For ISW3B = 0 (boundary layer neglected), FAREA is based upon the geometric dimensions of the nozzle or channel at the throat. For ISW3B ≠ 0, the displacement thickness at the throat is taken into account using equation I(125), I(129), or I(133) as appropriate. After it has been calculated in square centimeters, FAREA is converted into square feet. The total mass flow is then computed by multiplying FAREA by the sonic mass flux in lbm/ft²sec.

Finally, the flow conditions, including the calculated mass flow TFLØW, are printed out in the form of a list. This is followed by a list of species mole fraction.

3.37.2 Entry ØUT2

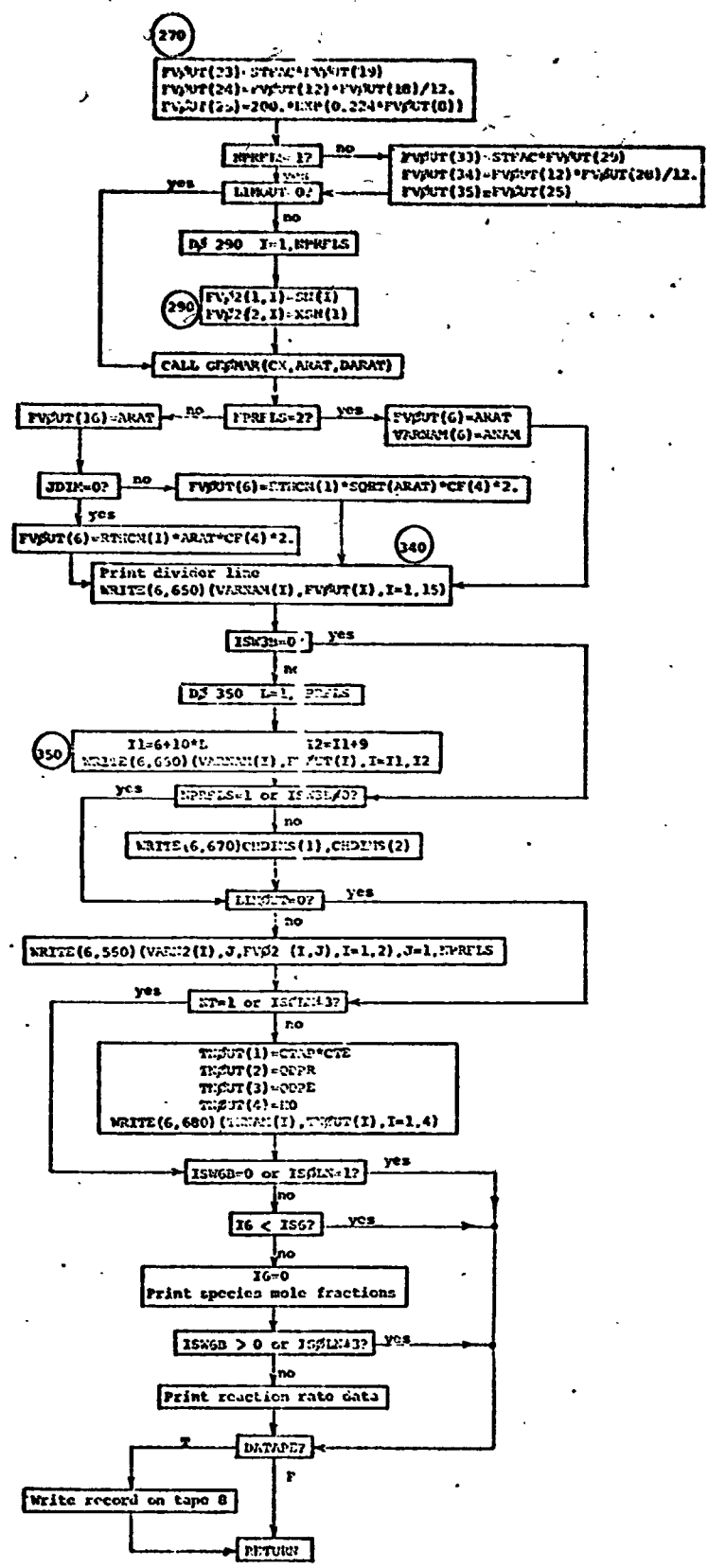
The operations performed when the entry ØUT2(ISØLN) is called, which are represented by a double box in figure 23, are flowcharted in detail in figure 24. The complexity of this portion of the subroutine results from the need to deal with various special cases:

ISW3B ≠ 0	Boundary layer included; output includes boundary layer displacement and momentum thicknesses, heat flux, shear stress, Stanton number, etc.
ISW3B = 0	Boundary layer neglected.
NPRFLS = 1	Flow is in a nozzle.
NPRFLS = 2	Flow is in a channel; two full sets of boundary layer outputs if ISW3B ≠ 0.
JDIM = 0	Two dimensional nozzle.



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Figure 24a. Flowchart of the Operations Produced by Calling Entry OUT2 of Subroutine OUT1 (Part a)



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Figure 24b. Flowchart of the Operations Produced by Calling Entry PUT2 of Subroutine PUT1 (Part b)

- JDIM = 1 Axisymmetric nozzle.
- LIMOUT ≠ 0 Include output of SN, XSN boundary
layer parameters.
- LIMOUT = 0 Omit this output.

The printed outputs are, in most cases, expressed in different physical units from those used in the internal computations of the code. Thus, a major function of OUT2 is to re-express these quantities in the desired output units. The converted flow variables in output units are loaded into an array $\text{FVOUT}(I)$, dimensioned (35). This scheme allows the use of compact WRITE statements based on implied DO loops. The converted quantities in FVOUT are also used in the calculations of conditions on wedge models; they are communicated to subroutine WEDGE through the common block OUTPUT . Definitions of the FVOUT array elements are given in the glossary of Fortran symbols, Section 4.31. However, the definitions will be reviewed here to facilitate the study of the flowchart (figure 24).

The following elements of FVOUT always have the meanings indicated:

- (1) Axial coordinate, x (inches)
- (2) Temperature, T ($^{\circ}\text{K}$)
- (3) Enthalpy, h (Btu/lb)
- (4) Pressure, p (atm)
- (5) Density, ρ (lbm/ft^3)
- (7) Velocity, u (ft/sec)
- (8) Mach number, M
- (9) Entropy, s ($\text{Btu}/\text{lbm}\text{-}^{\circ}\text{R}$)
- (10) Frozen specific heat ratio, γ

- (11) Effective area ratio, A_e
- (12) Reynolds number per foot
- (13) Molecular weight (g/mole)
- (14) Viscosity (lbm/ ft-sec)
- (15) Electrical conductivity (mho/cm)

Element number 6 is absent from this list because its meaning varies with the type of case, as explained below. For $ISW3B \neq 0$, the Reynolds number per foot, viscosity, and electrical conductivity are calculated in BLAYER, and ØUT2 obtains these data from common blocks /BLØUT/ and /TRPRØP/. For $ISW3B = 0$, BLAYER is not called and these quantities must be computed in ØUT2. The viscosity and conductivity are obtained by calling subroutine TRANSP.

If the boundary layer is included ($ISW3B \neq 0$), the ten elements FVØUT(16) to FVØUT(25) are loaded with data relating to the boundary layer. If, in addition, a channel geometry is being used ($NPRFLS = 2$), these elements contain the data for the boundary layer on the first profile, and the ten elements FVØUT(26) to FVØUT(35) contain the corresponding data for the second profile. The meanings for the elements numbered 17 to 25 are as follows:

- (17) Boundary layer displacement thickness (inch)
- (18) Boundary layer momentum thickness (inch)
- (19) Heat flux to nozzle wall (Btu/ft²-sec)
- (20) Shear stress on nozzle wall (lbf/ft²)
- (21) Recovery enthalpy (Btu/lbm)
- (22) Prandtl number at the reference temperature
- (23) Stanton number

(24) Reynolds number Re_0 based on momentum thickness

(25) Value of Re_0 for boundary layer transition

The data on geometric area ratio and nozzle diameter or channel widths is placed into $FV\emptyset UT(6)$, $FV\emptyset UT(16)$, and $FV\emptyset UT(26)$. The arrangement of these data depends upon the type of case being treated:

- A. If $ISW3B = 0$, $NPRFLS = 1$: $FV\emptyset UT(6)$ = nozzle diameter (inch). In this case, the geometric area ratio is equal to the effective area ratio, $FV\emptyset UT(11)$.
- B. If $ISW3B = 0$, $NPRFLS = 2$: $FV\emptyset UT(6)$ is set to a large value which exceeds the width of the F format used and thus prints as a row of asterisks. The width of the channel surface lying on the first profile is set to $CHDIMS(2)$. The corresponding width for the surface with the second profile goes into $CHDIMS(1)$. These $CHDIMS(I)$ values are printed on a separate line in the block of output. The geometric area ratio is again equal to the effective area ratio.
- C. If $ISW3B \neq 0$, $NPRFLS = 1$: $FV\emptyset UT(16)$ = geometric area ratio, $FV\emptyset UT(6)$ = nozzle diameter (inch).
- D. If $ISW3B \neq 0$, $NPRFLS = 2$: $FV\emptyset UT(6)$ = geometric area ratio; $FV\emptyset UT(16)$ = width of channel surface lying on the first profile (inch); and $FV\emptyset UT(26)$ = width of channel surface lying on the second profile (inch).

Note that in the channel case ($NPRFLS = 2$), the width of the surface lying on a given profile is twice the ordinate of the other profile.

After the data have been loaded into the output arrays, they are printed out. First, a line of asterisks is printed to separate the new block of output from the previous one. Imbedded in this line is a word ("frozen", "equilibrium", or "nonequilibrium") identifying the type of solution being

computed, based upon the entry-point argument ISØLN. In the case of a nonequilibrium solution, the number of integration steps since the last output of flow variables (NSTEPS) and the value of the indicator INEQ are also printed in this divider line. Note that NSTEPS is initialized in ØUT1 to the seven-digit value 1000000, so that when printed (with an I6 format) in the frozen and equilibrium solutions, it gives a row of six asterisks.

Following the divider line, the first fifteen elements of FVØUT(I) are printed by a single WRITE statement. The array VARNAM(I) appearing in this statement contains the Hollerith labels for the output variables. Then, if ISW3B is nonzero, the boundary layer data in FVØUT(16) to FVØUT(25) or FVØUT(35) are printed. For channel cases without the boundary layer, the channel widths CHDIMS(I) are printed on a separate line. If LIMØUT is nonzero, the boundary layer parameters stored in FVØ2(I,J) are printed on a separate line. For nonequilibrium solutions based on an electronic nonequilibrium model, the electron temperature, the radiative and electronic energy transfer terms, and the total enthalpy are loaded into TNØUT(I) and printed.

Next, if ISW6B is nonzero and ISØLN is not equal to 1, and if I6 is equal to IS6, the species mole fractions are printed. The test on I6 provides output of these species data every IS6th time ØUT2 is called in the equilibrium and nonequilibrium solutions, where $IS6 = |ISW6B|$. Finally, if ISW6B is negative and ISØLN = 3 (nonequilibrium solution) and if I6 = IS6, a table of reaction rate data is printed out. This table includes P_i , χ_i , and $P_i \chi_i$ for all of the reactions ($i = 1$ to r), and $d \ln \gamma_j / dx$ for all the species ($j = 1$ to n).

If DATAPE is .TRUE., a record is written on tape 8 for subsequent use by the NATA plot program.

3.38 Subroutine PERT

The function of subroutine PERT is to compute the perturbations in the flow variables by solving the following system of $n+2$ linear equations:

<u>Equation Index</u>	<u>Equation</u>
1 to c	I(368) for j = 1 to c
c+1 to n	I(362) for j = c+1 to n
n+1	I(374)
n+2	I(380)

The unknowns are $PERTGJ(J) = \delta \chi_j$ for $j = 1$ to n , $PCT = \delta T/T_0$ and $PRH\emptyset = \delta \rho / \rho_0$, in the order^j listed. The simultaneous solution is obtained by calling subroutine DSMS \emptyset L.

The coefficients and constants for the above system of equations are set up in the common array AA. First, the region of AA which is to be used is zeroed. Then, in the $D\emptyset$ loops ending at statement 20, the coefficients for equations I(368) are set into AA(I,J) for I = 1 to ISC. The coding down to the statement following 110 then loads the coefficients of equations I(362) into AA(I,J) for I = ISC + 1 to ISS, and loads the coefficients for equation I(374) into AA(ISSP1,J). In the calculation of the coefficients for I(362), the Fortran index I corresponds to j in I(362), J corresponds to k and K corresponds to i. The coding from $D\emptyset$ 120 down to the second statement below 130 sets up the coefficients for equation I(380) in AA(ISSP2,I). Finally, the loop $D\emptyset$ 140 loads the constant terms of equation I(362) into AA(I,ISSP3) for I = ISCP1 to ISS. The remaining equations are all homogeneous, so that their constant terms have been set by the initial zeroing of the AA array.

After DSMS \emptyset L has been called, the values obtained for the perturbations are retrieved from the column AA(I,ISSP3). Then the $\delta \chi_i$ values, denoted by SDCHI(I), are calculated from equationⁱ I(360).

3.39 Function PI \emptyset MEG

This routine provides the Maxwell-averaged momentum transfer cross section for collisions of electrons with

neutral species as a function of the electron temperature, TE. The cross section is calculated by linear interpolation in a table stored in common block /TNEQ/.

3.40. Subroutine PRØP

The function of this routine is to compute the flow conditions at a specified temperature in the frozen flow solution. Since PRØP makes use of species thermal properties, a call to PRØP must always be preceded by a call to THERM.

The technique used by PRØP has been discussed in Section 6.4 of Volume I. The coding is straightforward. As noted in Section 3.22.1, the flow variables are computed in nondimensional form. In particular, equation I(267) for the pressure is written in the form

$$ZP = \ln \left(\frac{P}{P_0} \right) = -\ln P_0 + \sum_{j=1}^n X_{j0} \frac{S_j^0}{R_0} - \sum_{j=1}^n X_{j0} \ln X_{j0} - \frac{W_0 S_0}{R_0} \quad (123)$$

The first sum on the right is evaluated in PRØP as

$$\sum_{j=1}^n X_{j0} \frac{S_j^0}{R_0} = W_0 \sum_{j=1}^n \gamma_{j0} S_j^0 = CMA \cdot S4 \quad (124)$$

The remaining three terms on the right in (123) are computed as $W_0 \cdot ENT$, where ENT has been calculated in FRØZ EQ before PRØP is first called:

$$ENT = - \sum_{j=1}^n \gamma_{j0} (\ln \gamma_{j0} + \ln W_0) - \frac{\ln P_0}{W_0} - \frac{S_0}{R_0} \quad (125)$$

3.41 Subroutine PRTA

This subroutine controls the output of the flow conditions and executes the calls to the MØDEL routine during

the nonequilibrium solution. It also calls subroutine BLCALL with the argument FINAL = .TRUE., to update the derivative of the boundary layer displacement thickness and provide permanent increments to the Cohen-Reshotko boundary layer integral. PRTA calls BLCALL even if the boundary layer is being neglected in the flow solution (ISW3B = 0), because the output and model routines need the species mole fractions, which are computed and loaded into common block /TEMPRY/ by BLCALL. For ISW3B = 0, the call to the boundary layer routine BLAYER is suppressed in BLCALL.

The step size in the nonequilibrium integration is often quite small, especially in the region upstream of the throat, where some of the reactions may still be nearly in equilibrium. To avoid the excessive output which would result if the flow conditions were printed out for every step in such regions of small step size, NATA prints the results of the nonequilibrium solution at intervals of TPRINT in the non-dimensional temperature CT. The control parameter TPRINT is preset to 0.01, and is under input control (with the input name TPRNTI). A counter (NSTEPS) is used to determine the number of steps computed between successive printouts of the flow conditions. This information is included in the printed output.

If the flow solution is being generated by the perturbation technique (INEQ = 0), the temperature perturbation PCT is added to CT before subroutine ØUT2 is called to produce the output. Then PCT is subtracted to restore CT to its equilibrium flow value.

3.42 Subroutine PUTQIN

The computation of the cross sections $\bar{\Omega}_{ij}(\ell, s)$ required in the transport property calculations for any particular conditions of gas temperature, pressure, and composition is carried out in the NATA code under the general control of subroutine PUTQIN. The subroutine selects data for each step of the computations from the edited cross section data in common block /TRANS7/, calls the appropriate subroutine to carry out the computations, and then returns the computed

(1) cross sections for all species pairs to subroutine TRANSP in the Q(K,I,J) array of common block /TRANS1/ (see the discussion of subroutine TRANSP). The argument X of subroutine PUTQIN is the partial pressure of electrons in the specified gas mixture, in atmospheres, a quantity which is needed in the computation of effective Coulomb cross sections from equations I(100).

Before beginning the cross section computations, subroutine PUTQIN initializes Q(K,I,J) to zero for all pairs of species included in the cross section computations.* If the first entry in the V array is positive, as it should always be for proper operation of the code, Q(1,2,1) is set to $0.8 Q_C$, where Q_C is the quantity defined by equation I(100b), for later use in the computation of effective Coulomb cross sections for the mixture. The remainder of the subroutine down to statement 240 then consists of a loop which goes through the steps L of the cross section computation one by one and accumulates the values of the cross sections for all species included in the computations in the array Q(K,I,J). In this computation, the index $K = KQ(L)$ indicates the option to be used in performing the Lth step of the computations (see Section 4.6 of Volume II), M is the location of the first parameter for the Lth step of the computations in the parameter array V, and LQ1 and LQ2 are the locations respectively of the first and last pairs in the IQ, JQ array to which the computations of the Lth step are to be applied. For each step L of the computations, subroutine PUTQIN determines the option which is to be used for that step from the KQ array and then calls the appropriate subroutine to carry out the computations for that option, as indicated in Table II. For the Coulomb cross sections ($KQ = 2$) however, these computations are carried out in subroutine PUTQIN itself using the previously computed value $0.8Q_C$.

*Note that the number N of species in the cross section computations, which is stored in common block /TRANS8/, is not necessarily the same as the number of species in the transport calculations which is stored in common block /TRANS5/.

TABLE II

SUBROUTINES USED IN CROSS SECTION COMPUTATIONS

Option KKQ	Subroutine Used	Type
2	PUTQIN	1
3	QEXP	1
4	QEX	3
5	QTAP	1
6	QREPP	1
8	QLJ	1
9	QSAME	1
10	QMIX	2
11	Q11	2
12	Q12	1
13	Q13	2
14	Q14	2

As indicated in Table II, the subroutines called by PUTQIN may be divided into three general types, based on the manner in which they return their output. The first and most commonly used type, 1, returns the three cross section values $\bar{\Omega}(1,1)$, $\bar{\Omega}(2,2)$ and $B^* \bar{\Omega}(1,1)$ computed for the step to subroutine PUTQIN in the three locations $\emptyset M(1)$, $\emptyset M(2)$, and $\emptyset M(3)$ respectively of the dimensioned variable $\emptyset M$. The loop from statement 190 through 200 of PUTQIN then adds these computed values to the previously computed cross sections $Q(K, IQ, JQ)$, for each pair of species in the IQ, JQ array to which the step applies. For the type 2 subroutines, on the other hand, the cross sections are loaded directly into the Q array by the subroutine which calculates them, and statements 190 through 200 in PUTQIN are bypassed. The reason for this difference is that for the type 2 subroutines the computed values of the cross sections for a species depend on the indices of the species, so that a single set of cross section values cannot be returned for all pairs of species to which the step applies.

The single type 3 subroutine in the code, QEX (charge exchange cross section), returns the computed cross section values for the step in the variable ϕM as do the type 1 subroutines. However, in this case, the coding of subroutine PUTQIN offers a choice, determined by the input values of V for the option, .s to whether the computed cross sections for the step will be added to the previous cross sections by transferring control to statement 190, as for a regular type 1 subroutine, or will be used to replace the previous values of $\bar{\Omega}^{(1,1)}$ and $B^* \bar{\Omega}^{(1,1)}$ by transferring control to statement 170 and bypassing 190.

The final section of coding between statements 210 and 240 of PUTQIN produces a voluminous dump of cross section data under control of the input parameter ISW8B. For $ISW8B \leq 0$, this dump is not produced. For $ISW8B > 0$, the dump is produced the first time PUTQIN is called and every ISW8Bth time thereafter. Note that PUTQIN is called once each time subroutine TRANSP is executed to compute transport properties. For each step in the cross section calculations, the dump includes a line of output giving the number of the step; the code number KKQ of the cross section option used in the step; the positions in the IQ, JQ arrays of the first (LQ1) and the last (NQ(L)) pairs of species for which cross sections are calculated in the step; and the position M of the first unused parameter in the V array. Below this line of indexing information, the calculated values of the averaged cross sections $\bar{\Omega}_{ij}^{(1,1)}$ for all pairs of species in the mixture are printed out as they exist at the conclusion of the step. Together with the initial edit of the cross section calculations (Section 3.9 of Volume II), this PUTQIN dump allows detailed checking of the operation of the cross section calculations for a new gas model, since the effect of each step in the calculations upon each pair cross section $\bar{\Omega}_{ij}^{(1,1)}$ is clearly shown. The routine could easily be modified to print $\bar{\Omega}_{ij}^{(2,2)}$ and $B^*_{ij} \bar{\Omega}_{ij}^{(1,1)}$, in addition.

3.43 Subroutine QCØ

Subroutine QCØUL computes the factor $0.8Q_C$ in the Coulomb cross section from equation I(100b). Inputs for this computation are the temperature T in °K and the

electron pressure $X = n_e kT$ in atmospheres. The computed value of $0.8Q_c$ in 10^{-16} cm^2 is returned in ØM1 . For $C \leq 0$, $\ln(f \Lambda)$ in I(100b) is set to 1.0 while for $C > 0$ it is calculated correctly from I(100c) and I(100d). To prevent overflow when $n_e = 0$, a small quantity ($\approx 10^{-50} T^4$) is added to X before carrying out the computations.

3.44 Subroutine QEX

Subroutine QEX computes the averaged cross sections $\bar{\Omega}(l, l)$ and $B^* \bar{\Omega}(l, l)$ in 10^{-16} cm^2 for a pair of species between which there is an exchange interaction (i.e., an interaction such as charge exchange in which the identities of the two species are interchanged) having a cross section of the form

$$Q_{\text{ex}} = (A - B \log_{10} v)^2, \quad (126)$$

where v is the relative velocity in cm/sec. The formula for the averaged cross sections $\bar{\Omega}(l, s)$ is then (ref. 9):

$$\bar{\Omega}(l, s) = 2 [A - \alpha_s B - \frac{1}{2} B \ln(T/W)]^2 + 2 \beta_s B^2 \quad (127a)$$

where

$$\alpha_s = \frac{1}{2} [\ln(4k/N_0) + \sum_{n=1}^{s+1} \left(\frac{1}{n}\right) - \gamma] \log_{10} e$$

$$\beta_s = \frac{1}{4} \left[\frac{\pi^2}{6} - \sum_{n=1}^{s+1} \left(\frac{1}{n^2}\right) \right] (\log_{10} e)^2 \quad (127b)$$

$\gamma = 0.577216 \dots$ is Euler's constant, and W is the molecular weight. Putting the numerical values into equations (127) and I(91) then gives the formulas used for $\bar{\Omega}(l, l)$ and $B^* \bar{\Omega}(l, l)$ in the subroutine. Since the cross section $\bar{\Omega}(2, 2)$ is not affected by exchange (ref. 10), subroutine QEX sets the contribution $\text{ØM}(2)$ to this cross section equal to zero.

3.45 Subroutine QEXP

Subroutine QEXP computes the averaged collision cross sections $\bar{\Omega}(1,1)$, $\bar{\Omega}(2,2)$, and $B^* \bar{\Omega}(1,1)$ in 10^{-16} cm^2 for a pair of species interacting according to the exponential potential $\phi = Ae^{-r/\rho}$, where r is the distance between the two species in \AA . These cross sections are obtained by linear interpolation in Monchick's tables (ref. 11) of α versus $I(1,1)$, A^* , and B^* , where

$$\alpha = \ln(A/kT)$$

$$I(1,1) = \bar{\Omega}(1,1) / (4 \pi \alpha^2 \rho^2) \quad (128)$$

and

$$A^* = \bar{\Omega}(2,2) / \bar{\Omega}(1,1),$$

which are stored in common block /TRANS4/. The table lookup and interpolation are carried out by a call to subroutine QINTRP. The coding assumes that there are 50 entries in these tables.

3.46 Subroutine QINTRP

Subroutine QINTRP is used by several of the other cross section subroutines for interpolating tabular data stored in common block /TRANS4/. In this interpolation, the independent variable is assumed to be stored in monotonically increasing order in the array TL, starting at location N1, and the corresponding values of the three dependent variables in the corresponding locations of the OMEGA1, ASTAR and BSTAR arrays. The argument N of the subroutine represents the number of data points to be included in the table, and A and B are respectively the values of the independent and dependent variables for the interpolation. The subroutine searches the data in the TL array for the specified value A of the independent variable and, if A is found to lie within the range of the table, carries out a linear interpolation of the tabular data to determine the values B(J) of the three dependent variables corresponding to the specified value A of the independent variable. If, on the other

hand, A lies outside the range of the table, the last two data points in the table are linearly extrapolated to determine the values of the dependent variable B and, if ISW8B \neq 0, a message is printed out indicating that an extrapolation of the data was required. This message gives the location in the TL array at which the extrapolation occurred, the desired value A of the independent variable, and the two values of the variable from the TL array used in the extrapolation.

3.47 Subroutine QLJ

Subroutine QLJ determines the collision cross sections $\bar{\Omega}(1,1)$, $\bar{\Omega}(2,2)$ and $B^* \bar{\Omega}(1,1)$ at temperature T for a pair of species interacting according to the Lennard-Jones 6-12 potential $\phi = 4 \epsilon [(\sigma/r)^{12} - (\sigma/r)^6]$ by linear interpolation in tables of $T^* = kT/\epsilon$ versus $\bar{\Omega}(1,1)^* = \bar{\Omega}(1,1)/\pi\sigma^2$, $A^* = \bar{\Omega}(2,2)/\bar{\Omega}(1,1)$, and B^* which have been computed for this potential by Monchick and Mason (ref. 12). These tables contain data for 37 reduced temperatures T^* from 0.1 to 100 and are stored in the arrays TL, OMEGA1, ASTAR, and BSTAR respectively in common block/TRANS4/, starting at location NI = 501 in the arrays.

3.48 Subroutine QMIX

Subroutine QMIX computes cross sections from the empirical mixing rule I(102) and adds them to the previously computed cross sections $Q(K,I,J)$ for a series of species pairs given in the IQ, JQ arrays in common block/TRANS7/. The arguments LQ1 and LQ2 of the subroutine indicate the positions of the first and last pairs in the IQ, JQ array to which these computations are to be applied.

To avoid computing the square roots required in equation I(102) several times in a single call to QMIX, these quantities are stored in the array

$$SQT(K,I) \equiv \frac{1}{2} \sqrt{Q(K,I,I)} \quad (129)$$

Upon each entrance to the subroutine, the values of SQT(1,I) are set to 0.0 for all species I, to indicate that they have not yet been calculated in the current call to QMIX. (Note that values of SQT cannot be saved over from one call to QMIX to the next, since the values of the cross sections Q(K,I,I) may have changed between the two calls.) The remainder of the subroutine then consists of a loop over the species pairs I, J which are to be included in the computation. For each pair, the subroutine checks the values of SQT(1,I) and SQT(1,J) to determine if they have been calculated previously in the loop, and, if not, computes the required values of SQT from (129) for K = 1, 2, and 3. The SQT values for the pair are then used in equation I(102) and the results added to the previously computed cross sections Q(K,I,J) to obtain the final cross section values for the pair.

Since the edited cross section data prepared by subroutine XSECT may contain steps using the option KQ = 10 which do not apply to any species pairs, a provision to bypass the cross section computations completely when IQ1 > IQ2 is included in the subroutine QMIX to allow for this possibility

3.49 Subroutine QREPP

Subroutine QREPP computes the averaged cross sections $\bar{\Omega}(1,1)$, $\bar{\Omega}(2,2)$, and $B^* \bar{\Omega}(1,1)$ in units of 10^{-16} cm^2 for a pair of species which interact according to an attractive or repulsive inverse power potential of the form $\phi = Ar^{-\eta}$, where r is the distance between the species in Å. The cross section formulas for this potential are (ref. 13):

$$\begin{aligned} \bar{\Omega}(1,1) &= \text{OMEGA1(ITL)} / (T^{2/\eta}) \\ \bar{\Omega}(2,2) &= \text{ASTAR(ITL)} * \bar{\Omega}(1,1) \\ B^* \bar{\Omega}(1,1) &= \text{BSTAR(ITL)} * \bar{\Omega}(1,1) \end{aligned} \tag{130}$$

where OMEGA1(ITL), ASTAR(ITL), and BSTAR(ITL) are input quantities given by equations I(105).

3.50 Subroutine QSAME

Subroutine QSAME sets the cross sections for a species pair equal to a constant multiple of those computed previously for another pair (I,J). The coding of the subroutine permits a different constant factor to be used for each of the three cross sections $\bar{\Omega}(1,1)$, $\bar{\Omega}(2,2)$, and $B^* \bar{\Omega}(1,1)$.

3.51 Subroutine QTAB

Subroutine QTAB obtains the values of the cross sections $\bar{\Omega}(1,1)$, $\bar{\Omega}(2,2)$, and $B^* \bar{\Omega}(1,1)$ for a pair of species at a given temperature T by linear interpolation in tables of $\bar{\Omega}(1,1)$, $\bar{\Omega}(2,2)$ and B^* versus temperature for the species which are included in the input. The tabulated data are assumed to be given in the arrays OMEGA1, ASTAR, BSTAR, and TL, respectively, in common block/TRANS4/, with the tables beginning at location N1 in the arrays and containing NL data points. An option is also provided to multiply the tabulated cross sections by an arbitrary constant factor A in the computations.

3.52 Subroutine Q11

Subroutine Q11 multiplies the previously computed cross sections for a series of species pairs by the temperature dependent factor $f(T)$ given by equation I(108). The subroutine computes the value of this factor at the specified temperature T from the input data for the subroutine and then calls subroutine Q14 with the appropriate inputs to carry out the actual multiplication of the cross sections by this factor.

3.53 Subroutine Q12

Subroutine Q12 computes the cross sections $\bar{\Omega}(1,1)$, $\bar{\Omega}(2,2)$, and $B^* \bar{\Omega}(1,1)$ for a species pair from the generalized mixing rule given by equation I(109). As with the other type 1 subroutines discussed above, it returns the computed cross sections to subroutine PUTQIN in the dimensioned variable OM.

3.54 Subroutine Q13

Subroutine Q13 computes the cross sections $Q(K1, I, J)$ according to equations I(110) and I(111) for a fixed value of $K1$ specified in the cross section data ($1 \leq K1 \leq 3$) and for a series of species pairs I, J given in positions LQ1 through LQ2 of the IQ, JQ arrays. The computed cross sections then replace the previous values for these cross sections in the $Q(K, I, J)$ array.

3.55 Subroutine Q14

Subroutine Q14 computes the cross sections $\bar{Q}(1,1)$, $\bar{Q}(2,2)$, and $B^* \bar{Q}(1,1)$ for a series of species pairs by multiplying the previously computed values of these cross sections by a constant factor, as indicated in equation I(112). The computed cross sections then replace the previous values for these cross sections in the $Q(K, I, J)$ array.

3.56 Subroutine RADIUS

This small routine performs geometric calculations that are required at several points in the computation of the boundary layer at the first flow point in subroutine BLAYER. The inputs to RADIUS are the type of nozzle geometry ITYPE, the axial coordinate X, and the profile index L. The quantities computed and returned to BLAYER are the profile ordinate R, the geometric area ratio AG, and the factor AGJ, which is proportional to r^{2j} in equation I(171).

For ITYPE = 1 (two-dimensional nozzle), R is obtained by calling the entry GMAR of subroutine GEOMAR. Then $AG = R/R0$ where $R0$ is the half-width of the throat gap, and $AGJ = 1$ since $j = 0$.

For ITYPE = 2 (axisymmetric nozzle), R is again obtained by calling GMAR, $AG = (R/R0)^2$, where $R0$ here is the throat radius, and $AGJ = AG$ since $j = 1$.

For ITYPE = 3 (rectangular channel), the ordinates $YZ(1)$, $YZ(2)$ of the two profiles are obtained by calling the entry GMAR2 of subroutine GEOMAR, $AG = YZ(1) \cdot YZ(2) / Y0Z0$ where $Y0Z0$ is the product of the profile ordinates at the throat, $R = YZ(L)$, and $AGJ = [YZ(M)]^2$, where $M = 2$ if $L = 1$ and $M = 1$ if $L = 2$.

3.57 Subroutine READ

READ is the input routine for NATA. It reads the input data, sets up the geometric description and gas model for the problem, and initializes most of the control parameters.

3.57.1 Input Operations

The subroutine begins by reading the input data for the current case. The routine contains four READ statements, of which the first two are always executed. The first of these statements reads a comment card into the array ACØM. This information is subsequently written out as part of a heading for the output for the case. The second READ statement reads data for the variables included in namelist INPUT. This is the main input operation for the code. Namelist INPUT contains all of the inputs to NATA except those required for defining chemical elements, species, and reactions (read under namelist EINPUT), and those required for calculating the transport cross sections of species (read under namelist TINPUT). When the code is run using the chemical and cross section data compiled into the BLØCK DATA routines, as it is normally, the statements READ(5, EINPUT) and READ(5, TINPUT) are not executed.

All of the variables in INPUT are printed out at the head of the output for the case by the statement WRITE (6, INPUT). If EINPUT and TINPUT are read, their variables are also printed out in namelist format. The reason for breaking the code inputs into these three groups is to minimize the number of pages of such output in normal usage, when EINPUT and TINPUT are not used.

3.57.2 Sonic Mass Flux Calculation

If ISW2B \leq 0, NATA determines the equilibrium conditions in the reservoir from data on the reservoir pressure or stagnation enthalpy and the total mass flow. In these cases, the sonic mass flux SMASS is required in the calculations to determine the reservoir temperature. This quantity is computed in READ from the total mass flow \dot{M} (denoted by FLØW) and the geometric dimensions of the throat. The coding treats three cases:

(1) Two-dimensional nozzle

Here \dot{M} and FLØW denote the mass flow per unit length of the throat gap. Hence

$$SMASS = \frac{\dot{M}}{D} \quad (131)$$

where D is the throat gap. Since SMASS is required in gm/cm²-sec while \dot{M} is in lb/in-sec and D in inches, a numerical conversion factor from lb/in²-sec to gm/cm²-sec is inserted on the right. This factor is 453.5924/(2.54)² = 70.3069.

(2) Axisymmetric nozzle

In this case, \dot{M} is the total mass flow in lb/sec, and

$$SMASS = \frac{4\dot{M}}{\pi D^2} \quad (132)$$

Insertion of the conversion factor (70.3069) gives a numerical coefficient of 4 x 70.3069/ π = 89.5173.

(3) Rectangular channel

The sonic mass flux here is

$$SMASS = \frac{\dot{M}}{D_1 D_2} \quad (133)$$

where D_1 and D_2 are the throat gaps of the channel's two profiles.

3.57.3 Automatic Air Model Selection

If the input AAMS (acronym for "automatic air model selection") is .TRUE, and IGAS is either 1 or 2, subroutine READ resets IGAS to either 1 or 2 on the basis of an enthalpy or temperature criterion. IGAS = 1 gives the large air model, including five ion species, which is required in cases

with high reservoir temperatures. IGAS = 2 gives the small air model which includes only a single ion (NO⁺) and which can be used at lower temperatures. If the reservoir temperature is read in (ISW2B > 0), the air model is selected to be IGAS = 1 or 2 according as the reservoir temperature CTAPI is greater than or less than 6000°K. On the other hand, if the reservoir temperature is to be determined from mass flow data (ISW2B ≤ 0), then its value is not yet known at the time when READ is executed. In this case, the air model selection is based on a stagnation enthalpy criterion of 8000 Btu/lb. For ISW2B < 0, the stagnation enthalpy is an input. For ISW2B = 0, the inputs are stagnation pressure and mass flow; in this case, the stagnation enthalpy is estimated from the Winovich correlation (ref. 14). The Winovich estimate of the stagnation enthalpy is computed from the formula

$$HWINØV = \left[\frac{136.7 p_0}{m_*} \right]^{2.52} \quad (134)$$

where p_0 denotes the reservoir pressure in atmospheres and m_* the sonic mass flux in g/cm² sec. The numerical factor on the right in (134) differs from the coefficient (280) in Winovich's equation because it includes a factor converting m_* from g/cm²-sec to lb/ft²-sec.

3.57.4 Initializations Based on Inputs

With few exceptions, the input variables of NATA are not altered by execution of the program. Also, the Namelist input system resets only those variables which are actually referenced in the input cards for a given case. Thus, when several cases are run in a single job, the input cards for the second and subsequent cases need contain data only for the variables which the user wishes to change, e.g., the pressure, mass flow, nozzle index, etc.

The input-determined variables which are not altered by program execution are included directly in the namelists INPUT, EINPUT, and TINPUT. However, some input-determined variables are altered by the program. In such cases, the variable which is altered and its corresponding input are given different names and thus are stored in separate

locations. The input name is the internal variable name (or a contraction thereof) with the letter I added at the end. At the cost of some increase in storage requirements, this arrangement protects the input variables against alteration during execution.

3.57.5 List of Test Sections

The data in the input array TSDIAM specify points in the flow at which special calculations are to be done. In the case of a two-dimensional or axisymmetric nozzle, the entries in TSDIAM are nozzle diameters at which model-condition calculations are to be performed. For a channel, the TSDIAM entries are specified channel widths at which free-stream flow and boundary layer calculations are to be performed.

Subroutine READ uses the data in the TSDIAM array to set up another array, TSAR, whose values are actually used by the program to select the points at which the special calculations are done. For a nozzle, the TSAR entries are geometric area ratio values. For a channel, the TSAR are widths in centimeters. In generating the TSAR array, READ rejects any TSDIAM values which are less than or equal to the throat diameter for the profile involved, and determines the number (NTS) of valid TSAR entries. All the entries of TSDIAM are preset (in BLOCK DATA) to the value 1×10^{20} . Thus, any values in excess of 1×10^{20} are assumed not to have been set by the user in input, and are not regarded as valid points.

3.57.6 Geometric Description

A nozzle is specified by one profile curvefit, a channel by two. For each profile, the parameters defining the geometry of the nozzle or channel may be either read in as part of the input or, if one of the standard geometries is desired, obtained from precoded data. For the Jth profile ($J = 1$ or 2), if $NPR\emptyset FL(J)$ is nonzero, the geometric data are looked up in the ZPRP array, which is set in BLOCK DATA. This array occupies the common block /NZZZ/. If $NPR\emptyset FL(i)$ is zero, the geometric data are set on the basis of the geometric input variables such as NSECTS, DIAM, and PARAMI.

In either case, after the geometric model has been set up, subroutine READ prints out a summary description of it, unless the standard geometry specified is the same as in the preceding case in the job.

3.57.7 Extra Chemical Elements

If additional chemical elements have been defined in the EINPUT input, the data for them are loaded into the EPRP array, which contains the precoded data for elements. The number of input-defined elements is NEELS. For the Ith such element, the atomic number is obtained from EEP RP(1,I) and the atomic weight from EEP RP(2,I). The index assigned to the element in the master list of elements is IEEP(I). The Hollerith atomic symbol for the element is obtained from the array ASYM, which is set in a data statement in READ.

3.57.8 Gas Model

The elements, species, and reactions present in the gas mixture are specified by the group of variables ISC, ISS, ISR, IC, IE(I), NSC, QPJ(I), JCS(I), IS(I), and IR(I). If IGAS=0, these variables are obtained from the input data. If IGAS is positive, they are looked up in the array of precoded data, GPRP(I,IGAS), which occupies common block /MIXT/. The physical definitions of the entries in this array are given in Section 4.5 of Volume II. If IGAS is negative, the input value is saved in the location IGS, and IGAS is reset to its own absolute value. In this case, all of the above-listed variables except QPJ(I) are again looked up in GPRP, but the QPJ (the mole fractions of the cold species) are obtained from input. This provision allows input control of the overall elemental composition of the standard gas mixtures containing more than one chemical element.

After these general specifications of the gas model have been determined, the required properties of the chemical elements, species, and reactions are looked up. The data for elements are obtained from the array EPRP(I,IE), where the second subscript is the index in the master list of elements. The species data are looked up in the array SPRP(I,IS), where IS is the index in the master list of species. The reaction data are obtained from RPRP(I,IR),

where IR is the index in the master list of reactions. The entries in these arrays are all defined in Section 4 of Volume II. The order of the elements, species, and reactions in a gas model is determined by the index arrays IE(I), IS(I), IR(I), and may differ from the order in the corresponding master lists.

3.57.9 Geometric Sequence of Model Points

Positions at which model condition calculations are to be done can be determined by specifying an initial point XMØDPI, a final point CXMAXI, and a number of points NMØDPT. From these data, subroutine READ computes a factor FACMP which is used to generate a geometric sequence of NMØDPT values of x extending from XMØDPI to CXMAXI. This factor is calculated from the formula

$$\text{FACMP} = \left[\frac{\text{CXMAXI}}{\text{XMØDPI}} \right]^{1/(\text{NMØDPT}-1)} \quad (135)$$

3.58 Subroutine RESET

This Fortran routine, used only in the IBM version of NATA, simulates the UNIVAC 1108 timing routines RESET and TIME. The time data are obtained by calling the library routine ACUCPU, which returns in its argument ICPU the number of milliseconds of CPU time remaining before the TIME parameter of the job step is reached. Subroutine RESET is called by the main program near the beginning of each NATA job. The initial value of ICPU is stored in IZERØ. Subsequent calls to the entry TIME then give the elapsed time I in milliseconds.

3.59 Subroutine RESTMP

RESTMP performs the iterative solution for the reservoir conditions when the second or third option for input specification of these conditions is used. These options have been described in Section 6.5 of Volume I. Usage of these options is controlled by the indicator ISW2B. For ISW2B = 0, the second option is used, and for ISW2B < 0, the third option. (For ISW2B > 0, the first option is used and RESTMP is not called.)

3.59.1 Second Option

The second option is implemented by the coding down to statement 90. In this option, the input specification of the reservoir conditions is based on the reservoir pressure, p_0 , and the total mass flow, \dot{M} . In the code, the value of p_0 in atmospheres is available in the location PRESA in unlabelled common, and the sonic mass flux m_* in $\text{g/cm}^2\text{sec}$ is given by SMASS in common block /MASSFL/.

The basic problem in the second option is to determine the reservoir temperature, T_0 , from these data. The code capabilities available to support this effort are

- (1) Entry INTA of subroutine INGAS, which computes all of the conditions in the reservoir from PRESA = p_0 and CTAP = T_0 .
- (2) Subroutine NRMAX, which computes the sonic mass flux m_* for equilibrium flow, based on given reservoir conditions.

RESTMP determines T_0 by an iteration in which INTA and NRMAX are called, at each step, and T_0 is repeatedly adjusted to make the calculated sonic mass flux nearly equal to SMASS.

RESTMP starts the iteration with a standard guess $T_0 = 10000^\circ\text{K}$. Subroutines INTA and NRMAX are then called to compute the reservoir conditions and sonic mass flux based on the trial value of T_0 . The call to INTA must be preceded by a call to subroutine INIT to set a number of species properties which are nondimensionalized using the reservoir temperature.

The sonic mass flux computed by NRMAX is in the non-dimensional form

$$SM = \frac{\rho_* u_*}{\rho_0 u_s} = \frac{m_*}{\rho_0 \sqrt{R_0 T_0} / \sqrt{W_0}} \quad (136)$$

From the perfect gas law,

$$p_0 = \frac{\rho_0 R_0 T_0}{W_0} \quad (137)$$

Elimination of ρ_0 between (136) and (137) gives the relation

$$T_0 = \frac{W_0(T_0)}{R_0} \left[\frac{p_c \cdot SM(T_0)}{m_*} \right]^2 \quad (138)$$

which is satisfied by the correct value of the reservoir temperature T_0 . In (138), W_0 denotes the reservoir molecular weight (CMA), R_0 the universal gas constant in mechanical units (8.31434×10^7 erg/mole $^\circ\text{K}$), p_c the reservoir pressure in dyne/cm², $SM(T_0)$ the nondimensional sonic mass flux computed by NRMAX, and m_* the sonic mass flux (SMASS, g/cm²sec) based on the input total mass flow \dot{M} . Equation (138) may be rewritten

$$T_0 = \frac{W_0(T_0)}{c} [SM(T_0)]^2 = \frac{F(T_0)}{c} \quad (139)$$

where the parameter c (denoted by CONST in the program) is a constant of the iteration:

$$c = R_0 \left[\frac{m_*}{p_c} \right]^2 \quad (140)$$

The second estimate of T_0 is obtained using equation (139), with W_0 and SM values based on the first trial value. The quantity SM is found to vary only weakly with T_0 , while the molecular weight is monotonically decreasing function of T_0 at constant pressure. Thus, the error in the second estimate of T_0 is opposite in sign to that of the first estimate; the two values bracket the correct solution. Accordingly, after INIT, INTA, and NRMAX have been called, the third estimate of T_0 is obtained by a linear interpolation technique. The function $F(T_0)$ is approximated as a linear function between $T_0 = T_1$ and $T_0 = T_2$ (the first two estimates). Then

$$\frac{T_0 - F_2}{T_0 - T_2} = \frac{F_2 - F_1}{T_2 - T_1} \quad (141)$$

Elimination of F between (139) and (141) gives

$$T_0 = - \frac{T_2 F_1 - T_1 F_2}{T_2 - T_1} \cdot \frac{1}{\frac{F_2 - F_1}{T_2 - T_1} - C} \quad (142)$$

for the third estimate.

Beginning with the fourth estimate, the dependence of $F(T_0)$ upon T_0 is approximated by a quadratic function based upon the data from the three most recent iterations:

$$F = D_1 + D_2 T_0 + D_3 T_0^2 \quad (143)$$

Substitution of the data from the three points gives a system of linear equations for D_1, D_2, D_3 . This system can be written in matrix form

$$A \cdot D = B \quad (144)$$

where

$$A = \begin{bmatrix} 1. & T_1 & T_1^2 \\ 1. & T_2 & T_2^2 \\ 1. & T_3 & T_3^2 \end{bmatrix} \quad (145a)$$

$$B = \begin{bmatrix} F_1 \\ F_2 \\ F_3 \end{bmatrix}, \quad D = \begin{bmatrix} D_1 \\ D_2 \\ D_3 \end{bmatrix} \quad (145b)$$

This system of equations is solved using subroutine SIMQ, which places the solution vector D into the locations originally occupied by the vector of constants B. Then elimination of F between (139) and (143) gives

$$T_0 = C (D_1 + D_2 T_0 + D_3 T_0^2) \quad (146)$$

The solution of this quadratic equation for T_0 is

$$T_0 = \frac{1}{2D_3} \left[- (D_2 - \frac{1}{C}) \pm \sqrt{(D_2 - \frac{1}{C})^2 - 4 D_1 D_3} \right] \quad (147)$$

The sign before the radical is determined by noting that, for $D_3 \rightarrow 0$, the solution must approach

$$T_0 = - \frac{D_1}{D_2 - 1/C} \quad (D_3 = 0) \quad (148)$$

Thus, the sign represented by \pm must be opposite to that of $-(D_2 - 1/C)$, i.e., the same as that of $(D_2 - 1/C)$.

In the program, the temperature values T_1, T_2, T_3 are stored in the array CTSAVE and the parameter values F_1, F_2, F_3 in the array F. If SIMQ returns an error indicator $KS \neq 0$, which will occur if the matrix (145a) is singular, RESTMP reverts to the linear interpolation technique. The convergence criterion for the iteration is that the relative change in T_0 between two successive iterative steps be smaller than or equal to 10^{-4} .

3.59.2 Third Option

The coding of the third option begins with statement 90 of RESTMP and continues to the end of the subroutine. In this option, the reservoir conditions are specified by input of the total mass flow \dot{M} and the stagnation enthalpy h_0 . The latter quantity is given, in calories per gram, by the variable HS in common /RDMAIN/. The third option requires a double iteration to determine both the reservoir temperature T_0 and the reservoir pressure p_0 from the available data.

To start the iteration, an initial estimate of the reservoir pressure is obtained from an approximate sonic flow analysis for a perfect gas. For a gas with constant specific heats,

$$p_* = \left[\frac{2}{\gamma + 1} \right]^{1/(\gamma - 1)} p_0 \quad (149a)$$

$$u_* = a_* = \sqrt{\frac{2}{\gamma + 1}} a_0 \quad (149b)$$

$$a_0 = \sqrt{\frac{\gamma p}{\rho}} = \sqrt{\frac{\gamma R_0 T_0}{W_0}} = \sqrt{(\gamma - 1) h_0} \quad (149c)$$

These relations are derived, for example, in Liepmann and Roshko (ref. 15, pp. 51-54). Combination of equations (149) and (137) gives

$$p_0 = \frac{\gamma - 1}{\gamma} \sqrt{\frac{\gamma + 1}{2(\gamma - 1)}} \left[\frac{\gamma + 1}{2} \right]^{1/(\gamma - 1)} m_* \sqrt{h_0} \quad (150)$$

In RESTMP, an initial estimate of p_0 is obtained from equation (150) using a standard estimate of the specific heat ratio, $\gamma = 4/3$. This value gives

$$p_0 = 0.743 m_* \sqrt{h_0} \quad (151)$$

in which all quantities are assumed to be expressed in absolute cgs units. With p_0 in atmospheres and h_0 in cal/gm, this relation becomes

$$p_0(\text{atm}) = 4.8 \times 10^{-3} m_* \sqrt{h_0(\text{cal/g})} \quad (152)$$

which is the form used in RESTMP.

The iteration scheme used to solve for the reservoir temperature and pressure is illustrated schematically in figure 25. The starting point for the iteration (labelled "1") lies at the pressure (152) and an initial estimate of temperature, 10,000°K. In the first stage of the double iteration, the temperature is varied at constant pressure to make the reservoir enthalpy equal to the input value HS (denoted by h_g in the figure). The coding of this stage runs from statement 110 down to (but not including) 130. At each step of this iteration on temperature, INIT and INTA are called, and the reservoir enthalpy is computed from CHA using equation I(241). In the first step, the temperature

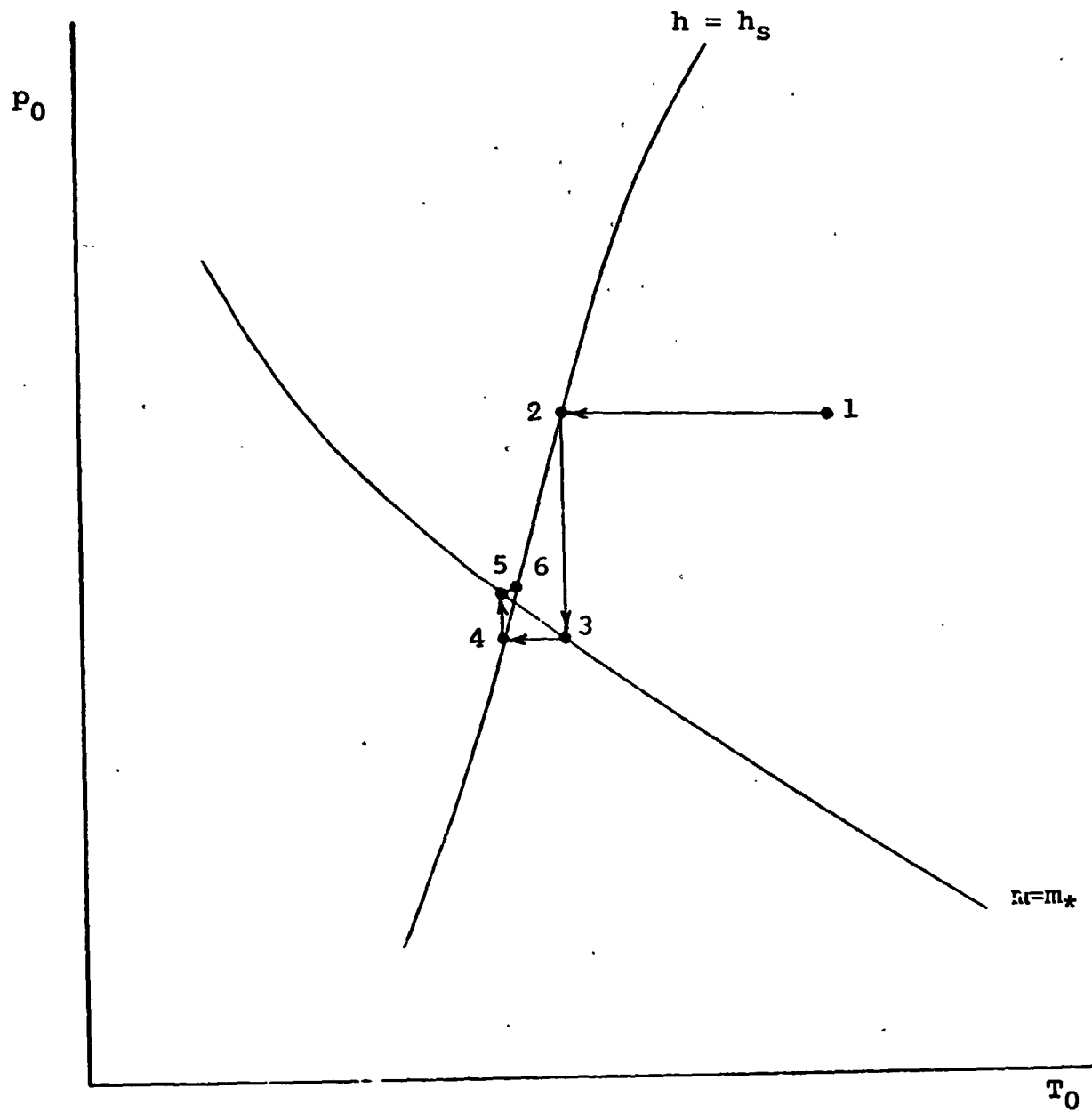


Figure 25. Iteration Scheme of Third Option for Specifying Reservoir Conditions

estimate is improved by assuming that the enthalpy is approximately proportional to the temperature:

$$T_0 = T_1 \cdot \frac{h_s}{h_1} \quad (153)$$

where T_1 is the initial estimate of T_0 and h_1 is the specific enthalpy corresponding to T_1 . In subsequent steps, the relation between T and H is approximated by a linear relation which does not necessarily pass through the origin:

$$T_0 = T + (h_s - h) (T_p - T) / (h_p - h) \quad (154)$$

This iteration on temperature is continued until the calculated enthalpy is equal to the desired value to within 1 part in 10^5 . The result of this stage is the point "2" in figure 25.

In the second stage, the reservoir pressure is varied at constant reservoir temperature to make the sonic mass flux equal to its desired value, SMASS. This stage is implemented by the coding from statement 130 down to (but not including) 150. In each cycle of this stage of the iteration, subroutine NRMAY is called to compute the non-dimensional sonic mass flux, SM. The sonic mass flux m^* in $g/cm^2 sec$ is calculated from this value using equation (136). The estimate of the reservoir pressure is then improved on the assumption that m^* is directly proportional to p_0 , and subroutine INTA is called to determine the reservoir conditions based on the new p_0 value. This stage is continued until the calculated sonic mass flux agrees with the desired value to within 1 part in 10^5 . The resulting point is "3" in figure 25.

The complete iteration consists of alternate temperature-adjusting and pressure-adjusting stages, and produces a series of points "4", "5", "6", etc., which spiral in towards the desired solution point in $T_0 - p_0$ space (figure 25). After each pair of stages (2,3), (4,5), etc., tests on temperature and pressure are applied to assess the convergence of the entire procedure. The criteria for convergence are that the relative changes in T_0 and p_0 during a stage-pair be less than or equal to 10^{-4} .

If the input control parameter ISW A is nonzero, a dump is written in each cycle of this iteration to allow tracing of the subroutine's operation.

3.60 Subroutine RNKT

RNKT is the routine that computes the changes in the dependent flow variables over an integration step using the Treanor-Runge-Kutta scheme explained in Section 7.5.1 of Volume I. Figure 26 is a flowchart of RNKT.

In the initializations, IFAIL is an indicator for the cause of step failure, used in diagnostic output which is produced when ISW5B is set to a negative value in the input (see glossary of Fortran symbols for common block /TNERK/ in Section 4). TE is a logical indicator. A .TRUE. value means that a chemical nonequilibrium model is in use. If TE = .FALSE., an electronic equilibrium model is being used. LIM denotes the number of dependent variables included in the numerical integration. For a chemical nonequilibrium model, $LIM = n + 1$ and the variables are γ_j for $j = 1$ to n and $CT = T/T_0$. For an electronic nonequilibrium model, $LIM = n + 3$ and the variables are γ_j , CT , $CTE = T_e/T_0$, and $CHA = h_0 W_0/R_0 T_0$. The energy transfer to the electron gas (QDPE) at the start of the integration step is denoted by QDPEB. Finally, DQMRK is the maximum allowable change in QDPE from QDPEB during the integration step.

For ISW5B different from zero, diagnostic dumps are produced at five locations in RNKT. The coding used to generate these dumps is omitted from the flowchart, as it is not essential to the function of the subroutine.

The computations in RNKT closely follow the analysis given in Section 7.5.1 of Volume I. The values y_1, y_2, y_3, y_4 are represented by arrays GJ1(J), GJ2(J), GJ3(J), GJ4(J), where J runs from 1 to LIM. Thus, these arrays contain not only the species concentrations but also the other dependent variables, CT, etc. The derivatives f_1, f_2, f_3 are represented by arrays F1(J), F2(J), F3(J). The parameter $(-P \Delta x)$ in I(398) is represented by an array P(J). The quantities F_1, F_2, F_3 defined by equations I(398) are represented by X1(J), X2(J), X3(J).

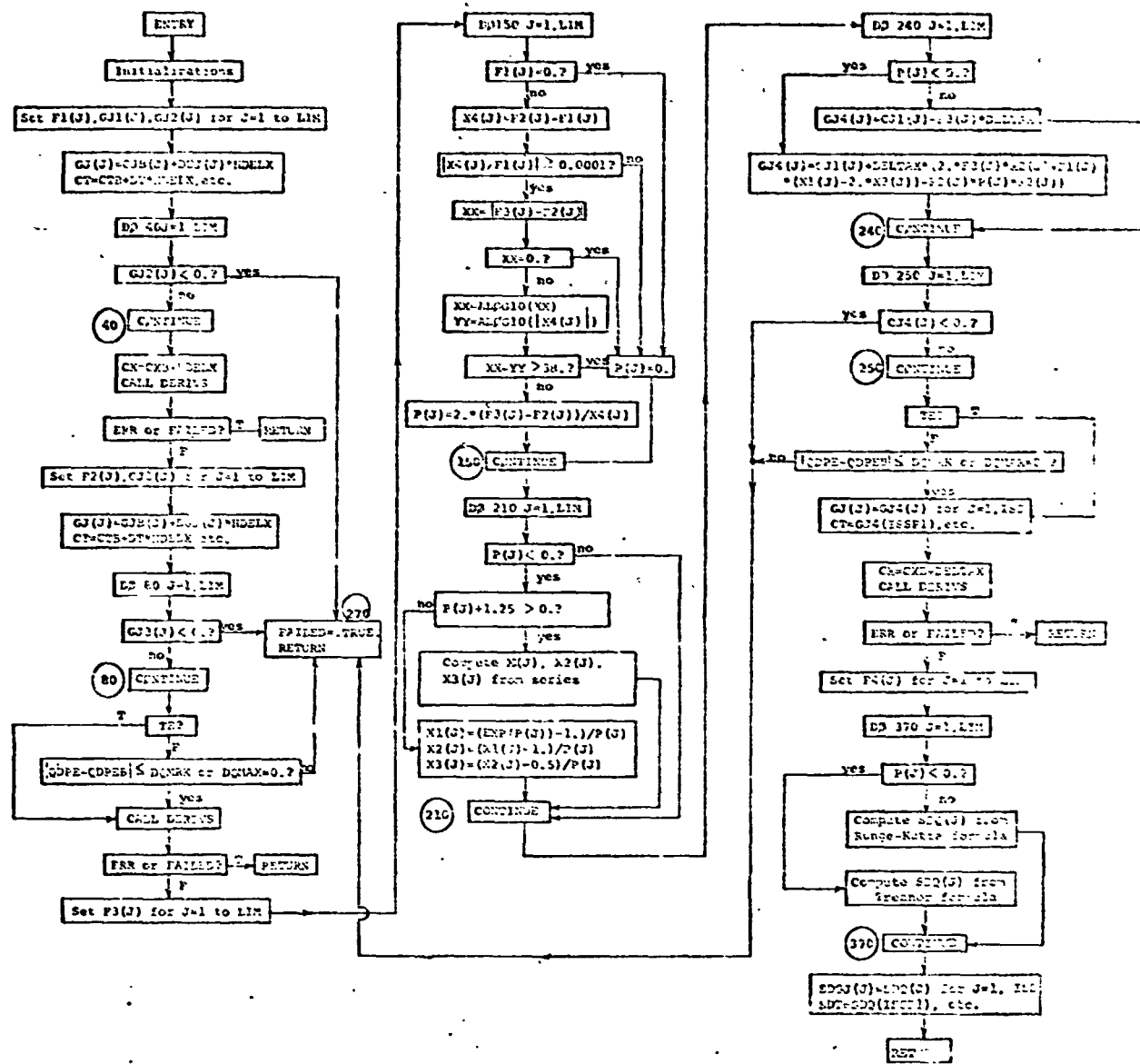


Figure 26. Flowchart of Subroutine RKKT

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Following the initializations, RNKT loads data into the F1, GJ1, and GJ2 arrays. If any of the GJ2 values is negative, FAILED is set to .TRUE. and the RETURN is executed. If not, CX is set equal to CXB + HDELX, where HDELX is one-half of the step size DELTAX. Then DERIVS is called to compute the derivatives F2(J), and the GJ3 array is set. The signs of all the GJ3(J) values are tested. In addition, if TE is .FALSE., the new value of the energy transfer QDPE to the electron gas, produced by the call to DERIVS, is tested to determine whether $|QDPE - QDPEB|$ is less than or equal to DQMRK. If not, the error return is executed unless DQMAX = 0 (which means that the current step is the first step in the numerical integration).

If the solution thus far passes these validity checks, DERIVS is called again to determine the derivatives F3(J). Next, the parameters P(J) are calculated. The basic formula for P(J) is I(402). In the Fortran notation, with $P(J) \rightarrow -P \Delta x$ in I(402), this equation becomes

$$P(J) = 2 \cdot \frac{F3(J) - F2(J)}{F2(J) - F1(J)} \quad (155)$$

However, under certain conditions, P(J) is set to zero. As explained in Section 7.5.1 of Volume I, this is done if $(f_2 - f_1)/f_1 = X4(J)/F1(J)$ has a magnitude smaller than 10^{-4} , to avoid excessive loss of accuracy on subtractions. P(J) is also set to zero if F1(J) = 0, to avoid a divide check error in the test just described, or if the evaluation of (155) would lead to a floating point overflow on the indicated division. For $P(J) \geq 0$, the changes of the dependent variables are computed using the Runge-Kutta formula I(392) in place of Treanor's formula I(401).

Once the P(J) have been determined, RNKT proceeds to compute X1(J), X2(J), and X3(J). If $P(J) \leq -1.25$, these quantities are calculated from equations I(398). Since the Fortran P(J) corresponds to $(-P \Delta x)$ in equations I(398), these formulas can be rewritten in the form

$$F_1 \equiv X1(J) = \frac{e^{P(J)} - 1}{P(J)} \quad (156a)$$

$$F_2 \equiv X2(J) = \frac{X1(J) - 1}{P(J)} \quad (156b)$$

$$F_3 \equiv X3(J) = \frac{X2(J) - \frac{1}{2}}{P(J)} \quad (156c)$$

which is used in RNKT. If $P(J) > -1.25$ (but negative), the subtractions in (156) could cause excessive loss of accuracy. In this case, F_1 , F_2 , and F_3 are evaluated from the power-series expansion

$$F_n = \sum_{k=0}^{\infty} \frac{[P(J)]^k}{(n+k)!} \quad (157)$$

which can easily be derived by substituting the Maclaurin expansion for e^x into (156). In RNKT, the sums (157) for F_1 , F_2 , F_3 are accumulated in the locations S_1 , S_2 , S_3 . The general terms in the three expansions are denoted by T_1 , T_2 , and T_3 , respectively. These terms are initialized to their values for $k = 0$, namely $T_1 = 1$, $T_2 = 1/2$, $T_3 = 1/6$. Then the successive terms in the series are generated recursively:

$$(T_n)_k = (T_n)_{k-1} \cdot \frac{P(J)}{n+k} \quad (158)$$

where $(T_n)_k$ denotes the k^{th} term in the expansion for F_n . In RNKT, the denominator $(n+k)$ in (158) is represented by $(n+Z-3)$, where Z is initialized to 3 for $k = 0$ and is incremented by 1 each time around the iterative loop. The

evaluation of (157) proceeds until the addition of T3 to S3 no longer affects the value of S3 to within floating-point accuracy.

Next, the GJ4(J) array is set. For each J, if P(J) is negative, GJ4(J) is computed from equation I(405). For zero or positive P(J), GJ4(J) is calculated from equation I(394c). After the usual validity checks, the flow variables GJ(J), CT, etc., are set to their corresponding values in GJ4(J), CX is set to CXB + DELTAX, and a final call to DERIVS is executed. The resulting values for the derivatives of the flow variable are loaded into F4(J), and the array SDQ(J) of changes in the dependent variables over the whole integration step is computed. For each J, if $P(J) \geq 0$, the Runge-Kutta formula I(392) is used to calculate SDQ(J). If $P(J) < 0$, then Treanor's formula I(401) is used. Finally, the results are transferred from the SDQ(J) array into the storage locations SDGJ(J), SDT, etc., used by other parts of the program.

3.61 Subroutine SHOCK

Subroutine SHOCK* computes the oblique shock angle and the conditions behind the shock for supersonic flow of a perfect gas over a wedge. The first three of the arguments are inputs to the routine. BMF denotes the free-stream Mach number M, DEL the flow deflection angle (i.e., the angle of attack of the wedge surface, α), and GAM the specific heat ratio γ . The outputs are FL and IERR, which are defined in Section 4.128 and by comment cards in the subroutine listing.

The method of solution is based upon the relation between the shock angle σ and the deflection angle α , which is given, for example, in Liepmann and Roshko (ref. 15, p.87). This relation can be expressed as a cubic equation in $\sin^2 \sigma$:

*This subroutine was programmed by Harvey Buss, Avco Systems Division. Statements for calculating quantities not required in NATA have been converted into "comments" to save storage.

$$\sin^6 \sigma + b \sin^4 \sigma + c \sin^2 \sigma + d = 0 \quad (159)$$

where

$$b = -\frac{M^2+2}{M^2} - \gamma \sin^2 \alpha \quad (160a)$$

$$c = \frac{2M^2+1}{M^4} + \left[\frac{(\gamma+1)^2}{4} + \frac{\gamma-1}{M^2} \right] \sin^2 \alpha \quad (160b)$$

$$d = -\frac{\cos^2 \alpha}{M^4} \quad (160c)$$

Equation (159) has three roots for $\sin^2 \sigma$, in general, of which the lowest corresponds to a decrease in entropy and thus to a physically impossible flow. The lower of the two remaining roots gives the desired weak-shock solution, while the uppermost root gives a strong-shock solution (e.g., a normal shock with $\sigma = \pi/2$ for $\alpha = 0$). Subroutine SHOCK first obtains the lowest of the three roots by solving equation (159) using Newton's method. With $u = \sin^2 \sigma$ and

$$F(u) = u^3 + bu^2 + cu + d \quad (161a)$$

$$F'(u) = 3u^2 + 2bu + c \quad (161b)$$

an improved estimate u' of the root is obtained, at each step of the iteration, using the formula

$$u' = u - \frac{F(u)}{F'(u)} \quad (162)$$

The convergence criterion is that $u' = u$ to within 1 part in 10^8 . If convergence is not achieved within 50 iterations, an error message is written and the RETURN is executed.

Use of an initial estimate $u = 0$ ensures convergence to the lowest root. Once this root u_1 has been obtained, the desired weak-shock solution is obtained by factoring the equation $F(u) = 0$:

$$(u-u_1) (u^2 + Au + B) = C \quad (163)$$

Equating the coefficients of corresponding powers of u in (161a) and (163) gives

$$A = u_1 + b \quad (164a)$$

$$B = -d/u_1 \quad (164b)$$

The two remaining roots of the cubic are then obtained by equating the quadratic factor in (163) to zero:

$$u = \frac{1}{2} [-A \pm \sqrt{A^2 - 4B}] \quad (165)$$

If the flow-deflection angle α is too large, the argument of the square root in (165) becomes negative and equation (159) has no physically valid solution. In this case, the error indicator IERR is set to 0 and the RETURN is executed. When $A^2 - 4B$ is positive, the lower of the two values (165), u_2 , is equal to $\sin^2 \sigma$ for the weak shock.

After the shock angle σ has thus been determined, subroutine SHOCK calculates the conditions behind the shock using perfect gas relations (ref. 15, p. 86). The only result required for use in NATA is the shock pressure ratio,

$$\frac{p_2}{p_1} = \frac{2\gamma M^2 \sin^2 \sigma - (\gamma - 1)}{\gamma + 1} \quad (166)$$

3.62 Subroutine SIMQ

SIMQ is a standard subroutine for solving systems of simultaneous linear equations. It was obtained from the IBM Scientific Subroutine Package (ref. /), and was modified slightly for use in NATA. The modifications consisted of converting it into a double precision routine and the addition of simple coding to allow SIMQ to handle the special case of one equation in one unknown. The only available documentation of SIMQ is that provided by comment cards in the source program.

3.63 Subroutine STUNTS

This routine performs two separate and unrelated operations. When called through the principal entry (STUNTS), it produces tables of species thermal properties as functions of temperature for the species in the specified gas model. When called through the entry STUNT2, it gives an edit of the transport property calculations, and optionally, a table and a punched deck containing averaged species cross section data.

3.63.1 Thermal Property Edit

The thermal property edit is produced by the DØ loop ending at statement 110. For each species J, the properties are computed at a preset series of temperatures TT(I) up to the input value of the reservoir temperature CTAP. These temperatures are as follows:

From 100°K to 1000°K at intervals of 100°K
 From 1000°K to 3000°K at intervals of 200°K
 From 3000°K to 10000°K at intervals of 500°K
 From 10000°K to 20000°K at intervals of 1000°K
 From 20000°K to 50000°K at intervals of 2000°K
 From 50000°K to 125000°K at intervals of 5000°K

1

The maximum number of temperatures is 74. (However, the thermo fits for the molecular species in the compiled-in gas models are useful only up to about 20000°K.)

The thermal properties are calculated by calling subroutine THERM for each temperature. THERM is programmed to calculate the properties using two different methods, the physical model and the thermo fit (Section 2.2 of Vol. I, ref. 1). The physical model is used for temperatures more than 500°K below the switchover temperature, CTMXX·CTAP. The thermo fit is used for temperatures more than 500°K above the switchover temperature. In a range 1000°K wide centered at the switchover temperature, a linear combination of the results from the two methods is used, to provide a continuous transition.

1

To obtain thermal properties based solely on the physical model, rather than on a mixture of the two methods, subroutine STUNTS first sets CTMXX to 1000, and calls THERM for each temperature. With this large value of CTMXX, all of the temperatures are more than 500°K below CTMXX times the reservoir temperature. The results of the thermal property calculations based on the physical model are stored in the array TPRØP(I,K,M) for M = 1. In the case of a nonstandard species for which physical model data have not been provided (ETAJ(J) = 0), the elements of TPRØP for M = 1 are all set to zero and the calculations based on the physical model are skipped.

For each species, if thermo fit data have been provided (IGJ(J) ≠ 0), CTMXX is set to -1000 and THERM is called again for each temperature up to the reservoir temperature, CTAP. With this negative value of CTMXX, all of the temperatures are more than 500°K above CTMXX·CTAP, so that the calculations in THERM are based solely on the thermo fit. The results are stored in TPRØP(I,K,M) for M = 2.

Finally, all of the results for the species are printed out. The properties computed and printed are

$$\text{TPR}\emptyset\text{P}(I,1,M) = \frac{\mu_j^{\circ} - H_{j0}^{\circ}}{R_0 T}$$
 (chemical potential in excess of the formation enthalpy, divided by $R_0 T$)

$$\text{TPR}\emptyset\text{P}(I,2,M) = H_j - H_{j0}^{\circ}$$
 (molar enthalpy in excess of the formation enthalpy, kcal/mole)

$$\text{TPR}\emptyset\text{P}(I,3,M) = C_{pj}$$
 (molar heat capacity, cal/mole $^{\circ}$ K)

$$\text{TPR}\emptyset\text{P}(I,4,M) = S_j^{\circ}$$
 (molar entropy, cal/mole $^{\circ}$ K)

Examples of the thermal property edit are shown in figures 19 to 46 of Volume II (ref. 2).

3.63.2 Transport Property Edit

The transport property edit consists of three parts, the first two of which are always obtained when STUNT2 is called. These parts are (1) an edit of the steps in the transport cross section calculation as specified in the input or in the precoded data; (2) an edit of the "edited" cross section calculations after unneeded steps have been deleted, and needed but unspecified steps have been added on the basis of default options; (3) an edit of the averaged pair cross sections as functions of temperature.

(1) Specified steps in the cross-section calculation.-- This first part of the transport property edit is implemented by the D \emptyset loop ending at statement 150. The index L runs over the steps of the cross section calculation in the order in which they are to be performed. The sequencing array ISEQ(L) selects the steps from the input or precoded data; i.e., the step which is carried out first is the one indexed I = ISEQ(1), etc. NNQI is the number of species pairs to which the cross section calculated in the Lth step will be applied. The indices of the species belonging to such a species pair are IIQJ, JJQJ (referred to the

master list of species).* For each step, STUNT2 prints out L, I, the index KKQ(I) specifying the type of cross section formula used in the step, the parameter values VV(K,I), and the species names for the first pair of species to which the results of the step are applied. The names of species for the remaining species pairs (if any) are then printed on successive lines.

(2) Steps in the edited cross section calculation.--
The DØ loop ending at statement 190 prints out a description of the steps in the "edited" cross section calculation which will actually be used in the transport property calculations for the current case. This "edited" calculation differs from the "input" calculation in two major ways: The steps applying only to species which are absent from the current gas model have been deleted, and steps which are required but which were not specified have been provided on the basis of default options** such as the cross section mixing rule.

The index L in the DØ loop ending at 190 runs over the steps in the edited calculation, in the order in which they are to be performed. KQ(L) is the index of the option (cross section formula) to be used in the Lth step. NV(KK) is the number of parameters used by this option. All of the parameters for all of the steps are stored, without gaps, in the singly dimensioned array V(I). The parameters for the Lth step occupy the locations from I = MV1 to MV2 in this array.

The indices of the species belonging to the species pairs to which the steps of the cross section calculation apply are stored, without gaps, in two arrays IQ(LQ), JQ(LQ). These indices IQ and JQ are referenced to the species list for the current gas model, rather than to the master list of species. The index LQ for the last species pair to which the results of the Lth step apply is NQ(L).

*See Section 4.2 of Volume II (ref. 2).

**See Section 4.6 of Volume II (ref. 2).

For each step L, STUNT2 first prints L, the option index KQ(L), and the parameter list V(I) for the option. Then it prints the names, HP(I), HP(J), for the first species pair using a carriage-control character + in column 1 to place these names on the same line. If there are any additional species pairs to which the step is applicable, their names are printed on successive lines.

(3) Averaged transport cross sections.--If ISWLB is negative, STUNT2 prints a table giving the transport cross sections for the various species pairs as functions of temperature. If ISWLB = -1, STUNT2 also punches these same data on cards for use by other computer programs. Some of the cross sections for charged species depend upon the electron partial pressure as well as the temperature. In STUNT2, the mole fractions SAVEC(I) used by the transport property routines are set to their values in the reservoir (for the current case), and are held fixed as the temperature is varied. The pressure is also held constant at its reservoir value. Thus, all of the cross sections printed are based upon the electron pressure in the reservoir.

The calculations and output are provided by the DØ loop ending at statement 230. For each species pair (I,J) with $I \leq J$, the temperature is varied from 1000°K to the reservoir temperature CTAP by intervals of 1000°K. At each temperature, subroutine TRANSP is called. With ISWLB < 0, TRANSP does not compute the viscosity, Prandtl number, etc., but does call lower-level transport routines to set up the cross section array Q(L,I,J), where I and J are the species indices and L runs from 1 to 3. These data are printed (and punched for ISWLB = -1), together with the temperature, species indices, species names, and a counter ICARD which serializes the punched cards. Their relation to the cross sections used in transport theory is (Section 3 of Vol. I, ref. 1):

$$\begin{aligned}
 Q(1,I,J) &= \bar{\Omega}_{ij}^{(1,1)} \\
 Q(2,I,J) &= \bar{\Omega}_{ij}^{(2,2)} \\
 Q(3,I,J) &= B_{ij} * \bar{\Omega}_{ij}^{(1,1)}
 \end{aligned}$$

They are expressed in units of 10^{-16} cm^2 .

3.64 Subroutine THERM

THERM is the routine which computes the species thermal properties: μ_j° (chemical potential at standard pressure), H_j (molar enthalpy), S_j° (molar entropy at standard pressure), and C_{pj} (molar heat capacity at constant pressure). In its internal calculations, NATA uses these properties in nondimensional form; the quantities actually computed in THERM are

$$\text{XMJAT}(J) = \frac{\mu_j^{\circ}}{R_0 T} \quad (167)$$

$$\text{SEJ}(J) = \frac{H_j}{R_0 T_0} \quad (168)$$

$$\text{SENT}(J) = \frac{S_j^{\circ}}{R_0} \quad (169)$$

$$\text{CCPJ}(J) = \frac{C_{pj}}{R_0} \quad (170)$$

where $J = j$ is the species index in the list of species for the current case. In these formulas, R_0 denotes the universal gas constant and T_0 the reservoir temperature.

When the subroutine is called through the entry THERM1, only the heat capacity is computed.

THERM provides two methods for calculating the species thermal properties: the "physical model" and the thermo fit. These techniques have been explained and discussed in Section 2.2 of Volume I (ref. 1). Data for the physical model are provided for all standard species. In the current version of NATA, thermo-fit data are provided only for the diatomic molecules and molecular ions in air and the planetary atmosphere models. The array IGJ(J) indicates whether thermo fit data are available for each species. If $\text{IGJ}(J) = 0$, there are no thermo fit data for the J^{th} species; in this case the physical model is used

for the species at all temperatures. If $ETAJ(J) = 0$, there are no physical-model data, and the thermo fit is used at all temperatures.

In the case of species for which both physical-model and thermo fit data have been provided, the physical model is used from low temperatures up to a nominal switchover temperature, $CTMXXI$, which is preset to $5000^{\circ}K$; and the thermo fit is used at higher temperatures. To prevent possible program failures due to discontinuities in computed species properties resulting from small mismatches between the physical model and thermo fit results, the switchover is spread over a temperature range $1000^{\circ}K$ wide, centered at $CTMXXI$. Equations I(58) are used to mix the results from the two techniques in this temperature range.

The temperature variable used by NATA in its internal computations is CT , a nondimensional temperature defined as T/T_0 . The corresponding nondimensional switchover temperature is $CTMXX = CTMXXI/T_0$. For species and temperatures for which the thermo fit is used, the species properties are computed from the formulas I(33), I(34), I(29), and I(31). Prior to the first call to $THERM$, the thermo fit coefficients $TFA(J) = a$, etc., have been nondimensionalized in subroutine $INIT$ so as to allow direct use of the nondimensional temperature CT in the thermo fit formulas.

The physical model calculations are done in two steps. The first step computes the translational, rotational, and vibrational contributions to the species properties, and the second step the contributions due to electronic excitation. The first step is coded separately for monatomic and diatomic species, on the one hand, and for linear triatomic species, on the other. In the case of monatomic and diatomic species, the calculations are done using equations I(50), I(51), I(54), I(56), and I(29). For linear triatomic species, equations I(52), I(55), I(57), and I(29) are used.

If the control parameter $INEQV$ is nonzero, the properties of molecular species are calculated on the assumption that the vibrational degrees of freedom are frozen at

the reservoir temperature. In this case, the formulas given in Section 2.4 of Volume I (ref. 1) are used to calculate the species properties.

3.65 Subroutine THRØAT

The function of THRØAT is to adjust the nozzle geometry in the downstream region so as to make the effective area ratio continuous between the upstream solution by the inverse method and the direct downstream solution. It performs this function by multiplying the rescaling factor RSA, which is used in GEØMAR, and which is initially 1.0, by the factor AG/S1. Here S1 is the geometric area ratio based on the specified nozzle geometry before the adjustment, and AG is the geometric area ratio calculated from the effective area ratio AFNX. If the boundary layer is not included in the solution; AG is equal to AFNX. If the boundary layer is included, AG is computed from AFNX by calling subroutine AGSØLN. Since boundary layer displacement effects upon the inviscid flow are neglected during the nonequilibrium solution by the inverse method, the rescaling also provides continuity of the effective area ratio when the coupling of the inviscid flow with the boundary layer is switched on following the call to THRØAT.

In addition to calculating RSA, THRØAT prints an informative message specifying the conditions at the current flow point, and resets the indicator IUPD from 1 to 0 to signify that the switch from the inverse method to the direct integration has occurred.

3.66 Subroutine TRANSP

Transport property calculations are performed in the NATA code under the general supervision of subroutine TRANSP. This subroutine contains two entries: TRANSX which is called to initialize the routines, and TRANSP(TTAB,P) which carries out the actual transport property calculations for any specified gas conditions.

The entry TRANSX is called prior to the start of actual transport property calculations, in order to inform the

routines of the species to be used in the calculations and to perform certain required initializations. The entry first sets the total number of species $N=ISS$ to be used in the transport calculations and then calls subroutine XSECT which selects cross section data for the species to be used in the calculations from the data available in the code and arranges them in a form convenient for the subsequent computations. Finally, TRANSX calculates a number of quantities depending on the species molecular weights which will be required in the subsequent transport computations, before returning control to the calling program. The quantities calculated here are

$$ZM1(I) = \frac{128 \times 10^{-24}}{9 \pi N_0 k^3} W_i = 2.8567 W_i \frac{\text{cm}^2 (\text{OK})^3}{\text{milliwatts}^2 (\text{A})^4} \quad (171a)$$

$$ZM2(I) = \frac{4}{15 N_0 k} \times 10^7 W_i = 0.032064 W_i \text{ g } ^\circ\text{K}/\text{joule} \quad (171b)$$

$$C(I,J) = \frac{8 \times 10^{-12}}{3k} \sqrt{\frac{2 W_i W_j}{\pi N_0 k (W_i + W_j)}} \quad \text{for } I \leq J \quad (171c)$$

$$C(I,J) = \frac{W_i - W_j}{W_i + W_j} \quad \text{for } I > J \quad (171d)$$

Once the gas species have been defined by calling TRANSX, the transport properties for any given gas condition are computed by calling TRANSP(TTAB,P), where TTAB is the gas temperature and P is the gas pressure in atmospheres. The species mole fractions are obtained from the array SAVEC(J) in common block /TEMPRY/. The first step in the computation is the calculation of the molar specific heat for each species in the mixture. This calculation is carried out by calling the entry THERM1 of subroutine THERM. Input to the calculation is by means of the common variable CT which specifies the ratio between the temperature TTAB for which the specific heats are to be calculated and the reservoir temperature CTAP. The nondimensional molar heat

capacities at constant pressure $W_j c_{pj}/N_0 k$ for each species in the mixture are then returned to subroutine TRANSP in the common variable CCPJ(J). These values are saved by the routine in the array DH for later use in computing the internal thermal conductivity from equation I(95), and are also used to compute the nondimensional frozen specific heat $CPT\phi T = W c_{pf}/N_0 k$ for the mixture, which is required in the calculations of the Prandtl and Lewis numbers, equations I(83) and I(84). In calculating the specific heats, it is necessary to save the previous values of CT and CCPJ in a temporary location and then restore them for later use in other portions of the flow field calculations.

After completing the specific heat calculations, TRANSP next calls subroutine PUTQIN which computes the collision cross sections $\bar{\Omega}_{ij}^{(1,1)}$, $\bar{\Omega}_{ij}^{(2,2)}$, and $B_{ij}^* \bar{\Omega}_{ij}^{(1,1)}$ in units of 10^{-16} cm^2 for all pairs of species in the mixture at the specified temperature and pressure. These cross sections are returned to subroutine TRANSP in the upper half ($I \leq J$) of the common arrays $Q(1,I,J)$, $Q(2,I,J)$, and $Q(3,I,J)$ respectively. If an edit of pair cross sections has been requested ($ISWLB < 0$), a RETURN is now executed, and subroutine STUNTS then prints the cross sections contained in the Q array. However, in a normal NATA run ($ISWLB \geq 0$), subroutine TRANSP uses these values of the $\bar{\Omega}_{ij}^{(l,s)}$ to compute the matrix elements $A_{ij}^{(\alpha)}$, $a_{ij}^{(\alpha)}$ and $\Delta_{ij}^{(1)}$, equations I(88) to I(90), which are required in the transport property calculations, and stores them in the Q array, replacing the $\bar{\Omega}_{ij}^{(l,s)}$ values which were previously stored there. The statements down to 80 compute and store these quantities as follows:

$$Q(1,I,J) = \frac{1}{k} \sqrt{T} \Delta_{ij}^{(1)} \left(\frac{\text{cm} (\text{°K})^{3/2}}{\text{milliwatt}} \right) \quad \text{for } I \leq J \quad (172a)$$

$$Q(1,I,J) = \frac{25}{4} \sqrt{T} a_{ij}^{(K)} \left(\frac{\text{cm} (\text{°K})^{3/2}}{\text{milliwatt}} \right) \quad \text{for } I > J \quad (172b)$$

$$Q(2,I,J) = \frac{25}{8} \sqrt{T} ZM2(I) A_{ij}^{(\mu)} \left(\frac{\text{cm} (\text{°K})^{3/2}}{\text{milliwatt}} \right)$$

$$= \frac{5}{6} \frac{\sqrt{T}}{k} \Delta_{ij}^{(2)} \quad \text{for } I \leq J \quad (172c)$$

$$Q(2, I, J) = \frac{25}{8} \sqrt{T} a_{ij}^{(\mu)} \left(\frac{(^{\circ}K)^{1/2}}{\text{millipoise}} \right) \text{ for } I > J \quad (172d)$$

$$Q(3, I, J) = \frac{25}{8} \sqrt{T} A_{ij}^{(K)} \left(\frac{\text{cm } (^{\circ}K)^{3/2}}{\text{milliwatt}} \right) \quad (172e)$$

In connection with these computations, it may be noted that the diagonal matrix elements $a_{ii}^{(\alpha)}$ are not required in the transport calculations, as shown by equation I(86), so that it is unnecessary to compute them here. Since the quantity (172e) is not symmetric in i and j , it requires the whole $Q(3, I, J)$ matrix for storage, whereas the other matrix elements are all symmetric, and can be stored in half the array. The quantity $\rho c_p f D_{ij} / \sqrt{T}$ in milliwatts/cm $(^{\circ}K)^{3/2}$ is also computed and stored by the code as FLEWIS, where D_{ij} represents the binary coefficient I(99) between the two species ID1 and ID2 which have been specified for the calculation of the Lewis number in the code input.

The calculation of the transport properties from the matrix elements (172) is begun in the DØ loop ending at statement 90, and is then carried forward in a series of three subroutines ELCØND, KINT, and KANDMU. The quantities returned to TRANSP by these subroutines are

$$\begin{aligned} \text{SIGMA} &= \sigma \text{ in mhos/cm} \\ \text{ZKINT} &= K_{\text{int}} / \sqrt{T} \quad (\text{milliwatts/cm-} (^{\circ}K)^{3/2}) \\ \text{ZK(1)} &= K_{\text{tr}} / \sqrt{T} \quad (\text{milliwatts/cm-} (^{\circ}K)^{3/2}) \\ \text{ZK(2)} &= \mu / \sqrt{T} \quad (\text{millipoise/} (^{\circ}K)^{1/2}) \end{aligned} \quad (173)$$

where σ , K_{int} , K_{tr} , and μ are respectively the electrical conductivity, the internal thermal conductivity, the translational thermal conductivity, and the viscosity of the mixture, as given by equations I(85) to I(87), I(95) and I(98). These quantities are then combined further by subroutine TRANSP according to equations I(83), I(84) and I(94) to obtain the viscosity VISC in poise, and the frozen

Prandtl number PRF and the atom-molecule Lewis number* FLEWIS. These quantities, along with the electrical conductivity SIGMA, are then returned to the calling program in the common block TRPROP. For ISW8B \neq 0, these data, as well as certain other quantities of interest, are also printed out by the final section of subroutine TRANSP before returning control to the calling program.

3.67 Subroutine WEDGE

Subroutine WEDGE calculates the pressure, heat flux, and boundary layer displacement thickness on wedge models. The calculations are done using the modifications I(501), I(503), I(505) of the results of the Cheng-Kemp theory. Optionally, the calculations are also done using the unmodified Cheng-Kemp theory if the input control variable ISW9B is set to a value in the range from -1 to -3 .

WEDGE obtains the free-stream flow properties required in the calculations from the array FVØUT in common block /ØUTPUT/. The data in this array refer to the current flow point, because the free stream conditions are always printed out immediately prior to a call to subroutine MØDEL which, in turn, calls WEDGE. These data are accessed by equivalencing the variable names used in WEDGE to the corresponding FVØUT array elements.

The subroutine logic is straightforward. All quantities which are independent of the angle of attack and the wedge leading-edge radius are computed first. The remaining calculations are performed and the results are printed out in a pair of nested DØ loops over the index IR for leading edge radii and the index IA for angles of attack.

*The Lewis number as given by I(84) is multiplied by an integer variable IAMBIP, which is equal to 1 if the two species for which the Lewis number is calculated are both neutral, and equal to 2 if one of these species is an ion. The factor of 2 takes account of the ambipolar nature of the diffusion of ions in a neutral plasma.

For each angle-radius combination, the subroutine performs calculations for all of the specified distances from the leading edge. These distances are specified in the input in two independent ways:

- (1) A uniform sequence of NWX points, with an initial value of WX1 and an increment DWX;
- (2) A list of arbitrary values in ascending order.

Subroutine WEDGE folds these two sequences together into a single sequence arranged in ascending order, using logic similar to that employed in subroutine NEXTMP.

WEDGE calls subroutine SHOCK to compute the static pressure ratio across the oblique shock, denoted by APR(IA). This quantity is used in calculating the angle of attack parameter Γ (denoted by CAPGAM) from equation I(501) in the modified Cheng-Kemp theory. After Γ and ζ (ZETA) have been computed, WEDGE calls subroutine WESOLN to obtain the approximate solution of the Cheng equation. The arguments of WESOLN are double precision.

The calculations based on the modified Cheng-Kemp theory are performed for ITH = 1. When these calculations have been completed at a given model point, if ISW9B is negative, ITH is set equal to 2 and all of the calculations are done over using the unmodified theory.

The input variable ISW9B is also used to control the output of some results of secondary interest. If IS9 = |ISW9B| is equal to 1, the shock ordinate YS is printed; if IS9 is equal to 2, the nondimensional coordinate ZETA is printed; and if IS9 is equal to 3, both YS and ZETA are printed.

3.68 Subroutine WESOLN

The function of subroutine WESOLN is to solve the Cheng equation I(480) for z , zz' , and $(zz')'$ when ζ and Γ are given. The solution is based on the approximation I(484) which relates the solution for general Γ to that for $\Gamma = 0$.

The latter is obtained from Cheng's analytical solution I(482), I(483) for $\zeta > 10^{-8}$. For small ζ , there is a great deal of cancellation of terms in I(482), so that many significant figures of accuracy are lost. This problem is kept under control by performing all of the arithmetic in WESØLN in double precision, and by using an analytical series solution for $\zeta \leq 10^{-8}$:

$$z = 1.65096 \zeta^{2/3} + 0.50869 \zeta^{5/6} - 0.0249\zeta \quad (174a)$$

$$zz' = 1.81711 \zeta^{1/3} + 1.25975 \zeta^{1/2} + 0.14712 \zeta^{2/3} \quad (174b)$$

$$(:z')' = 0.6057 \zeta^{-2/3} + 0.62988 \zeta^{-1/2} + 0.09808 \zeta^{-1/3} \quad (174c)$$

Equation (174a) was derived by Boger and Aiello (ref. 16). Equations (174b) and (174c) can be obtained from (174a) by differentiation.

For $\zeta > 10^{-8}$, equation I(482b) is solved for λ using Newton's method. The parameter λ is denoted by XL. An initial trial value of λ is obtained from the rough analytical approximation

$$\lambda = zz' = (6 \zeta)^{1/3} + (3 \zeta)^{1/2} \quad (175)$$

which has a maximum error of about 12 percent. At each step in the Newton iteration, ζ is calculated from the current trial value of λ using equation I(482b), and an improved estimate λ' of λ is obtained from

$$\lambda' = \lambda - (\zeta - \zeta_i) \frac{d\lambda}{d\zeta} \quad (176)$$

in which ζ_i is the input value ZETA and, from I(480) with $\Gamma = 0$ and I(483),

$$\frac{d\lambda}{d\zeta} = \frac{d}{d\zeta} \left(z \frac{dz}{d\zeta} \right) = \frac{1 + \sqrt{zz'}}{z} = \frac{1 + \sqrt{\lambda}}{z} \quad (177)$$

where z is computed from I(482a). The convergence criterion of the iteration is that the ζ from I(482b) equal the input value to within 0.01 percent.

Once the correct value of λ for the solution with $\Gamma = 0$ has thus been found, the three outputs of the routine are computed as follows: zz' (denoted by ZZP) is obtained from equation I(491), $(zz')'$ (denoted by ZZPP) from I(434) and I(485), and z (denoted by Z) from I(492).

3.69 Subroutine XSECT

Subroutine XSECT provides NATA with the capability to perform transport property calculations for gas models which do not contain all of the species listed in the precoded cross section data, and in which the species may be listed in a different order. For this purpose the subroutine sets up a new set of data for the cross section computations in common block /TRANS7/ which gives the same values for the cross sections as the original input data in common block /TRANS2/, but which is in a more condensed format, and is referenced to the set of species included in the current gas model rather than to the larger set of species for which cross sections are defined in the precoded and input data. The edited data in common block /TRANS7/ are then used by the code in computing collision cross sections for the gas in the subsequent transport property calculations. The original cross section data are also retained unchanged in common block /TRANS2/, in order to allow the running of stacked cases with different sets of species.

Since several of the options which are used for computing cross sections in NATA (see Section 4.6 of Volume II, ref. 2) give values for the cross sections which depend on the values computed in previous steps of the computation, the edited cross section data in common block /TRANS7/ are set up to carry out the steps of the cross section computations in the same order in which they were specified in the

original data; however, the species references in the data are changed so that the computed cross sections will be indexed in the same way as the species used in the current gas model, rather than being indexed to the original precoded or input data. Further, cross sections for pairs of species not required in the transport calculations are omitted from the edited data, and any missing cross sections are supplied according to the default options described in Section 4.6 of Volume II. The procedures used by subroutine XSECT in preparing these edited data are described in the following paragraphs. It may be noted that several of the cross section options available in the code contain features which require special treatment in this process, so that the coding of the subroutine becomes rather lengthy, although the basic concept is simple. A schematic flowchart of the subroutine showing the major control sections is provided in figure 27.

3.69.1 Initializations

The first executable statement in subroutine XSECT sets the number of species N for which cross sections are to be computed equal to the number of species ISS in the current gas model. The next two statements set $IELEC = 1$ if electrons are to be considered in the transport calculations and $IELEC = 0$ otherwise. If electrons are included in the calculations, they must always be the first species.

The statements from "JJ = 50" down through statement 30 establish the correspondence between the species in the current gas model and the master list of species,* which is used in indexing the cross section data supplied in common block /TRANS2/. These statements use the common variable $IS(L)$, which gives the index IS in the master species list of the L^{th} species in the current gas model (for $1 \leq L \leq ISS$), to prepare a new array $I(J)$ which gives the position I in the current model of the J^{th} species in the master species list, for each species included in the current gas model. Species in the master species list which are not included in the current model are filled into the $I(J)$ array in order following the species included in the current model, so that

*See Section 4.2 of Volume II (ref. 2).

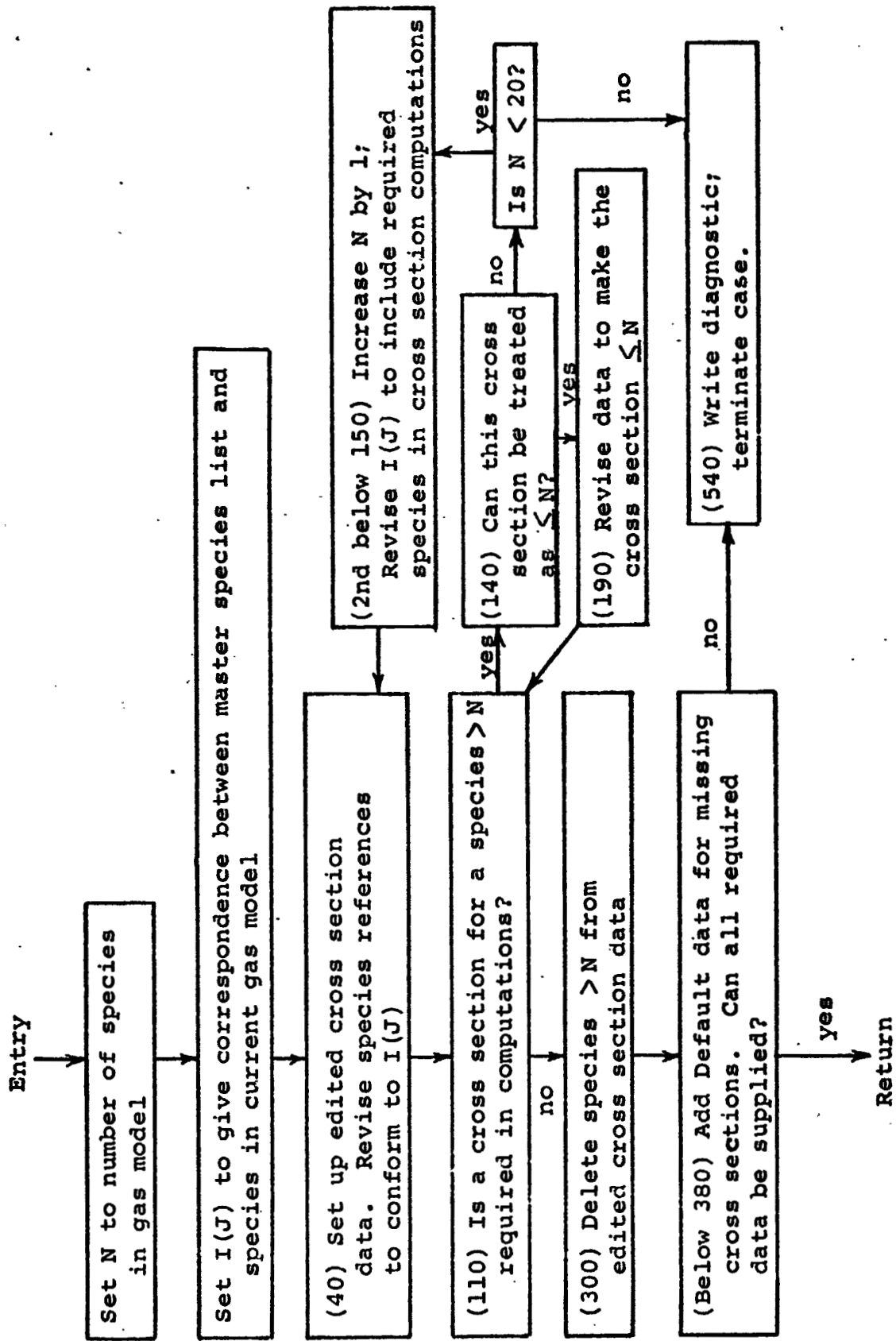


Figure 27. Schematic Flowchart of Subroutine XSECT

I(J) defines a rearrangement of the master species list such that all species J in the original master species list occur at some point I(J) in the rearranged list,* and the first ISS species in the rearranged list are just the species in the current model, in the correct order.

The three statements following statement 30 determine the indices ID1 and ID2 in the rearranged species list of the species to be used in the Lewis number calculations. They are set up so that ID1 < ID2 in all cases.

The common variable Q(1,J,L) is used as a flag in subroutine XSECT to indicate those pairs of species for which cross section data have been supplied. Statements 40 through 50 initialize this variable to 0 for all pairs of species included in the cross section calculations, and the value is then later set to 1 for each species pair for which data are provided. The value Q(1,2,1) = 1. is set initially to indicate that the pair (2,1) is used in the intermediate calculation of the cross sections in subroutine PUTQIN, and is thus not available for use in input. (Note that we are concerned here with ordered pairs (J,L). Only the pairs with $J \leq L$ are used in the actual transport property calculations,** while those with $J > L$ are available for intermediate calculations if desired.)

3.69.2 Resequencing of Steps and Revision of Species References

The coding following statement 50, down to statement 110, revises the species references in the input cross section data to conform to the rearranged species list I(J) described above and sets up a preliminary version of the edited cross section data in common block /TRANS7/ using these revised data. At this stage of the computation, the

*It is assumed that the master species list contains no more than 50 species; in the current version of NATA, array dimensions limit the number of species to 30.

**See the discussion of subroutine TRANSP in Section 3.66.

edited data contain the same information as the original cross section data in common block /TRANS2/, but the order of steps in the edited data is rearranged to conform to that specified by the sequencing array ISEQ, and the data on species pairs and cross section parameters for each step are close packed in the IQ, JQ, and V arrays. Further, $NQ(L)$ in the edited data is taken to be the sum from 1 to L of the original $NNQ(L)$ values, and the indices of all species pairs in the edited data are changed to refer to the positions of the species in the rearranged species list, $I(J)$. Since only cross sections $Q(J,K)$ with $J \leq K$ are used in the transport property calculations, the correspondence between species pairs used in setting up the edited cross section data must preserve this property. Accordingly, the pair (J,K) in the original data is replaced by $(I(J), I(K))$ if $J-K$ and $I(J)-I(K)$ have the same sign and by the pair $(I(K), I(J))$ if they have different signs. Since this correspondence is 1-1 between ordered pairs of species, it is clear that when every species pair in the original cross section data is replaced by its corresponding pair in the edited data, the calculated cross section values will all be the same as before, but will be rearranged in the order required for the transport property calculations.

The above correspondence between species pairs is carried out in the NATA code by subroutine CXSECT, which returns the revised indices of a species pair in its argument list when it is called with the original indices. In the coding down through statement 80, this correspondence is used to set up the IQ, JQ list of species pairs for the revised cross section computations. This section of coding also sets $Q(I,L,J) = 1.$ for those species pairs which are included in the IQ, JQ list.

For the cross section options $KQ = 9$ and 12 , there are also references to species pairs in the V parameters of the option which must be revised. This revision is carried out by the statements following statement 80 down through the statement before 110. Subroutine BXSECT searches through the steps L of the cross section computation in common block /TRANS7/, starting with the first step after the value of L specified in its argument, and looking for

a step with $KQ(L) = 9$ or 12 . When such a step is found, the subroutine returns the index of the step L , the index $MMV + 1$ of the first V parameter for the step, and the indices LL and J of the species pair which is referenced in the V parameters of the step. Subroutine $XSECT$ then calls $CXSECT$ to revise the indices for the pair and goes on with the search by calling $BXSECT$ again. This process is continued until the final step $L = NKQ$ of the computations is reached; at this point $BXSECT$ sets L to 0 as a signal that the search has been completed and control passes to the next section of code (statement 110). In addition to revising the indexing of the species pairs, this section of code also sets MMV , for later use, equal to the total number of V parameters used in the cross section computations.

3.69.3 Deletion of Unused Steps

When control reaches statement 110 of subroutine $XSECT$, the rearrangement of the species in the cross section data to conform to the order in the current gas model has been completed. The following section of coding, from statement 110 through 380, is now concerned with increasing the efficiency of the cross section computations by omitting unnecessary steps in the computation. In this procedure, the statements from 110 to 300 determine which cross sections are to be omitted from the computations, while those from 300 through 380 carry out the actual editing of the cross section data to delete these values.

As noted previously, the number of species N to be included in the cross section computations is set equal to the number of species in the current gas model at the beginning of subroutine $XSECT$. Thus only cross sections between pairs of species with indices $I \leq N$ in the revised cross section data are required in the transport property calculations. Ordinarily cross sections involving a species with index $> N$ can be omitted from the cross section computations without changing the values of the computed cross sections for the remaining species with indices $\leq N$; however, for the options $KQ = 9$ and 12 used in the cross section computations, it may happen that the cross section values

for some pair of species with indices $\leq N$ are to be computed from values obtained previously for some other species pair involving a species with index $> N$. Statements 110 to 140 check for this possibility, again using subroutine BXSECT to search for those steps in the computation with $KQ = 9$ or 12 . (Note that the coding here assumes that $L = 0$ when statement 110 is reached. This is always true, since statement 110 can only be reached from the IF statements following statements 100 and 290.) If this search does not reveal any pair of species with an index $> N$ which affects the cross section computations for species with indices $\leq N$, then it is possible to omit all species with indices $> N$ from the cross section computations without affecting the values of the calculated transport properties, and control passes at once to statement 300. If, on the other hand, some species pair (LL, J) with an index greater than N is found which affects the computations for indices $\leq N$, then the search for further such pairs is temporarily discontinued and the code immediately begins looking for a way to include the pair (LL, J) in the computations. The first approach considered is to treat the cross sections for the pair (LL, J) as an intermediate step in the computation of the cross sections for the species with indices $\leq N$, and to fill the computed cross sections for this pair into some unused location (II, JJ) of the cross section array for species $\leq N$, where $JJ < II \leq N$.* Statements 140 to 150 look for such a location, using the previously set array $Q(1, II, JJ)$ as a flag to indicate which locations are available.

If a suitable location for intermediate storage of the pair (LL, J) is not found in statements 140 through 150, then the following section of code down to statement 190 increases the number N of species to be included in the cross section computations by one, rearranges the list of species $I(J)$ for the edited cross section data so that the species corresponding to the maximum index of the pair (LL, J) will be the new species included in the cross section computations (note that by definition this species was not included

*Remember that only species with $II \leq JJ$ are used in the transport calculations.

in the computations previously), and then returns control to statement 40 to begin the preparation of the edited cross section data again from the beginning, using the new arrangement of species I(J). If this procedure would lead to a value of $N > 20$, however, so that the dimensions allowed for the cross section computations in the code would be exceeded, it is not carried out and control passes instead to statement 540 where a diagnostic message is printed and the error indicator ERR is set to terminate the case. If more than 50 species were specified in the master species list, the case may also be terminated by reaching the $G\emptyset T\emptyset 540$ statement after statement 180.

When the statements from 140 through 150 are successful in finding an unused location (II,JJ) which can be used for storing the cross sections of the species pair (LL,J), control passes to statement 190, where the value of $Q(1,II,JJ)$ is immediately set to 1.0 to indicate that the pair (II,JJ) is now being used in the cross section computations. The following statements down to statement 300 then revise the edited cross section data in common block /TRANS7/ to change all references to the pair (LL,J) to (II,JJ). To carry out this revision without changing the values of the computed cross sections, any steps in the computation which use the pair (LL,J) in the empirical mixing rule $KQ = 10$ must also be converted in the revision process to use the generalized mixing rule $KQ = 12$. This conversion is carried out by the coding down through the statement $NKQ=NSV$ after statement 270. The code first searches through all steps L of the cross section computations to find any steps using the option $KQ = 10$ which refer to the pair (LL,J). This may occur in two ways: either the species pair (LL,J) may be a non-diagonal pair (i.e., $LL \neq J$) which is computed with the option $KQ = 10$ at some step of the computations, or it may be a diagonal pair (i.e., $LL = J$) which is used in the option $KQ = 10$ to compute the cross sections for some non-diagonal pair involving the species LL. Both of these possibilities are checked for by the IF statement below statement 200, and if either is found the following statements then revise the edited cross section data in common block /TRANS7/ to convert the computation for the given species pair to the option $KQ = 12$. This is accomplished

by inserting the required data for the $KQ = 12$ option into the KQ , NQ , IQ , JQ and V arrays just ahead of the data for the step L which originally computed the cross section, and moving the remaining data in these arrays down to make room for the new data. The original option $KQ = 10$ is retained in the data immediately following the new $KQ = 12$ option, but is no longer applied to that species pair for which the $KQ = 12$ option is being used. The values of MMV , NKQ and L are also adjusted to agree with the revised data. In these computations, subroutine $BXSECT$ is used to locate the position MM in the V array at which the new data are to be inserted. This is done by replacing NKQ by L in common block $/TRANS7/$, so as to cause $BXSECT$ to end its search of the KQ array, when step L is reached. The true value of NKQ is saved in the variable NSV throughout this section of the code and is restored to its proper location at the end of the section.

After the required conversions from $KQ = 10$ to $KQ = 12$ have been made, control passes to the section of code between card $XSE 213$ and statement 300 , which changes all references to pair (LL, J) in the IQ , JQ , and V arrays to refer to the pair (II, JJ) instead, using the same algorithms as were used previously in rearranging the species in statements 40 through 110 . When these changes have been completed, control returns again to statement 110 and the code again searches the edited cross section data for a pair of species with an index $> N$ which affects the cross section computations for species with indices $\leq N$. If such a pair is found, it is eliminated from the edited cross section data as described above, and the whole process is repeated until either all species pairs with indices $> N$ which affect the cross sections for species $\leq N$ have been eliminated from the data and control passes to statement 300 , or else the procedures used in the code are unable to eliminate a species pair and cause the case to be terminated as described above.

By the time control reaches statement 300 in subroutine $XSECT$, the edited cross section data have been put into such a form that all cross section computations involving species with indices greater than N can be omitted

without changing the values of the computed cross sections for the species with indices $\leq N$. The revision of the data to omit these computations is carried out by statements 300 to 380. The pairs of species for which cross sections are computed at each step of the computations are given in the IQ and JQ arrays, so that the steps to be deleted from the computations can be determined simply by searching the IQ, JQ list for pairs in which at least one of the indices IQ or JQ is $> N$. This search is performed by the preliminary loop ending at statement 310, and the value of IQ for each pair which is to be deleted is set to 0 as a signal for the subsequent calculations.

To determine the required revision of the KQ, NQ, and V arrays in the edited data, it is necessary to establish the correspondence between these arrays and the IQ, JQ arrays. This is done and the arrays are revised in the loop over the steps L of the cross section computation which ends at statement 370. For each step L of the computations, the small loop ending at statement 320 determines the correspondence to the IQ, JQ list and calculates the total number of species pairs NEWLQ which are not deleted up through step L. This number is the new value of NQ for the step.

The second statement below 330 is the test to determine whether the current step L of the computations will be omitted from the revised data. The test compares the current value of NEWLQ with the value for the previous step; if these two values are equal, it indicates that the current step of the computations will not apply to any species pairs in the IQ, JQ list after the deletions are made, so that the step can ordinarily be omitted from the revised data without affecting the computed cross sections. However, for the option, $KQ = 10$ the step is not omitted, even though it does not apply to any species, since it may be required subsequently in supplying the default data for the cross sections, as described below.

The edited cross section data for those steps which are not to be omitted from the computations are placed in their proper locations in the revised KQ, NQ, and V arrays by the coding from the third statement below 330 down to the

statement before 360. Since the revised arrays in this stage of the calculations are always shorter than, or at most equal to, their initial length, this revision can be carried out entirely within the arrays themselves, without disturbing subsequent steps in the revision. The indices L and MV here indicate the positions of the data in the initial arrays, while NEWL and NEWMV are their positions after the desired omissions have been made. It may be noted further in connection with this calculation that no data need be added to the revised V array for $KQ = 10$, since this option requires no values for V in its operation. Also the number of V parameters $NV(K)$ required by each option is obtained from preset data in the code. For steps which are omitted from the computations, of course, no data need be added to the KQ, NQ, and V arrays; however, the total number of steps NKQ in the computations is reduced by 1 by statement 360.

Once the revision of the KQ, NQ, and V arrays has been completed, the revision of the IQ and JQ arrays is completed in the loop ending at statement 380 by deleting the unwanted species pairs and moving the remaining pairs up to fill in any resulting gaps. This process completes the editing of the cross section data to remove unnecessary steps in the computations which was begun at statement 110 of the subroutine, and provides a set of edited cross section data which will give the same computed cross sections as the original input data for all species included in the current gas model, but which includes only data for the N species for which cross sections are to be calculated, and omits all data for other species which may have been included in the input or the precoded data.

3.69.4 Default Options

The final section of subroutine XSECT, following statement 380, is for the purpose of providing cross section data according to the default options discussed in Section 4.6 of Volume II (ref. 2) for any pairs of species included in the transport calculations for which data were not supplied in the input. These data are added to the edited cross section data in common block /TRANS7/ to provide a complete set of data for the later cross section computations in subroutine PUTQIN.

The first section of coding below statement 380 supplies data for computing the unspecified neutral-neutral cross sections in the gas mixture from the empirical mixing rule, equation I(102). These data are added to the edited cross section data at the last step in which the option KQ = 10 is used, or, if this option is not used at all in the data, after the last step of the specified data. This procedure has been adopted in the code to permit data on the resonant contributions to the neutral-neutral cross sections to be specified after the last step in the computations using KQ = 10, without affecting the values of the default cross section computed from the mixing rule I(102).

The coding from statement 380 down to statement 400 determines the step LL in the cross section computations at which the unspecified neutral-neutral cross sections are to be added to the data. If no step with KQ(LL) = 10 is found in the data, an additional step is added for this purpose at the end, and the value of NQ(LL) for this step is set to NQ(LL-1), since as yet, the step does not apply to any pairs of species in the IQ, JQ list. The statements from 400 through 420 then determine the charge IZ, in units of the electronic charge, for each of the species in the transport calculations, using data from the common array LPIJ containing the α_{ij} matrix, and test the value of IZ to determine which are the neutral species. When no electrons are present in the gas, (i.e., IELEC = 0), all species are assumed to be neutral and IZ is not set. Since the correspondence I(J) which was previously established between the master species list for the code and the species in the transport property calculations is now no longer needed, the array I is now used to store the indices of the neutral species for later reference. Since no default option has been established in the code for the computation of cross sections for the collisions of a neutral species with itself, the code tests each neutral species to determine whether these data have been specified in the input and if they have not, terminates the case by transferring control to statement 540.

Once the neutral species have been determined, the code calls subroutine AXSECT to add the required data for these species to the edited cross section data in common block /TRANS7/. Subroutine AXSECT then takes the list of neutral species stored in the I array and tests all pairs (J,K) with $J \neq K$, using the previously set values of the common variable $Q(1,J,K)$, to determine those pairs for which cross section data have not previously been supplied. All such pairs found in this search are then added to the IQ, JQ list at the LL^{th} step of the cross section computations, the values of the $Q(1,J,K)$ for the added pairs are set to 1.0, and the IQ, JQ, and NQ lists are appropriately revised to take account of these additions. In these calculations subroutine AXSECT assumes that the species to be tested are stored in numerical order in the I array, and that $NQ(LL)$ has been properly set prior to entrance into the subroutine.

After completing the specification of the neutral-neutral cross sections, the code next goes on to add data specifying the unspecified ion-ion cross sections to the edited cross section data. The three statements immediately preceding statement 430 establish a location in the edited data for the insertion of these cross sections. If the last step NKQ in the current version of the data is not presently being used in the cross section computations, it is used for the new data, while if it is being used, an additional step is added to the computations at the end of the current data for the insertion of the new data.

The loop from statements 430 to 480 adds data for computing effective Coulomb cross sections $I(100)$ for the unspecified ion-ion collisions to the edited cross section data. The coding of this loop is set up to cause control to pass through the block of statements from 440 to 460 five times. For the case in which electrons are present ($NELEC = 1$), the values of the indices at each passage

through this block are shown in Table III.* For each passage through the block, the loop ending at statement 450 stores some of the species included in the transport property calculations in the array I, the exact species which are chosen depending on the values of the indices for that passage. Thus, referring to Table III, the array I consists of all singly charged ions in the transport calculations for the first passage through the loop, all singly charged ions plus electrons for the second passage, and so forth. The call to XSECT following statement 450 then finds all pairs of the species in the I array for which cross sections have not been previously specified, and revises the edited cross section data to include these pairs in the NKQ^{th} step of the computations. After returning to subroutine XSECT, the code checks to see whether any unspecified pairs were actually found by subroutine AXSECT in this search, and, if they were, supplies values of the parameters KQ and V for the step which are appropriate for the effective Coulomb potential $I(100)$ from the preset array VCØUL, and then adds another new step at the end of the computations to prepare for storage of the next set of cross section data. The number of entries NEWMV in the V array, which was set originally in the loop ending at statement 340, is also revised to conform to the new data. Thus, the first passage through statements 440 to 460 sets the cross sections for all unspecified pairs of singly-charged ions to the effective Coulomb cross sections given for singly-charged ions in equations $I(100a)$. The

*Note that for IELEC = 0, all species are assumed to be neutral so that all the unspecified cross sections were previously set in treating the neutral-neutral interactions. Thus, no cross sections remain to be set in the present loop and the edited cross section data are accordingly unaffected by the loop, independent of the values of the indices.

TABLE III

INDICES USED FOR CALCULATION OF EFFECTIVE COULOMB
CROSS SECTIONS IN SUBROUTINE XSECT

Passage number through loop	J	KK	JJ	IE	Species in I array*	Species pairs for which cross section data are supplied*
1	1	1	1	1	1	(1,1)
2	1	1	2	0	e,1	(e,e) (e,1)
3	2	2	1	1	2	(2,2)
4	2	1	1	1	1,2	(1,2)
5	2	1	2	0	e,1,2	(e,2)

*The notation e indicates electrons, 1 indicates singly charged ions, and 2 indicates doubly charged ions.

second passage then sets the cross sections for all unspecified pairs involving either singly-charged ions or electrons; but, since the ion-ion cross sections were already set in the previous passage, only electron-electron and electron-ion cross sections are actually set in this passage. From equations I(100a) one sees that the effective Coulomb cross sections for electron-electron and electron-ion collisions are the same (for singly charged ions), so that both types of cross sections can be set in a single step of the computations. Similarly, the last three passages through the loop set the unspecified cross sections involving doubly-charged ions to the values indicated in equations I(100a), as shown in Table III. Thus, when all five passages through the loop have been completed, cross sections will have been specified in the edited cross section data for all pairs of charged species to be included in the transport calculations, up through doubly ionized species. Further, it should be noted that the last step NKQ in the edited data at this stage of the calculations will be an empty position which is not used in specifying the cross sections but is available for the addition of further data.

The preset coding of subroutine XSECT does not provide any default option for specifying the cross sections of ions which are triply charged or higher. Thus the resection of code following statement 480 down through the IF statement below statement 490 checks for unspecified ion-ion cross sections for species with higher than double ionization, and, if any are found, terminates the case. Otherwise, the next three statements below the IF statement add all so far unspecified pairs of species in the transport property calculations to the final step NKQ of the cross section computations. Since all neutral-neutral and ion-ion cross sections have previously been specified, the cross sections added at this step are necessarily all neutral-ion cross sections. If any such cross sections are added, parameters are then added to the KQ and V arrays to compute them from an inverse fifth power potential ($KQ = 6$), using the data on the magnitude of the cross sections stored in the array locations OMEGA1(996), ASTAR(996), and BSTAR(996) in common block/TRANS4/. For the preset data used in the code, this results in the cross sections for the added pairs being set equal to the corresponding $N-O^+$ cross

sections. On the other hand, if no unspecified neutral cross sections are found, these data are not added and the value of NKQ is reduced by one to eliminate the unused step at the end of the cross section computations.

The above calculations complete the preparation of the edited cross section data in common block /TRANS7/. The number of steps NKQ in the computations is now checked, and if it does not exceed the maximum allowed by the program dimensions (NKQ = 100), control is returned to subroutine TRANSP. Otherwise the case is terminated by a transfer to statement 540.

In their final version, the edited cross section data prepared by subroutine XSECT specify the procedures to be used in computing all of the cross sections required for the transport property calculations in a given case, and are in a form suitable for later use by subroutine PUTQIN and its associated subroutines in the actual computation of these cross sections. These data are in their final form when control returns from subroutine XSECT, and are not altered by the code in the subsequent transport property calculations until XSECT is called again to prepare a new set of edited cross section data for another case.

4. GLOSSARY OF FORTRAN SYMBOLS

This section presents a complete glossary of Fortran symbols for the NATA code. Symbols appearing in common are listed first, followed by the remaining symbols in individual routines. The order is as follows: unlabelled common, labelled common in alphanumeric order of the block names, main program, subroutines and functions in alphanumeric order. The variables in each common block are listed in the order in which they appear in the block. The non-common variables are listed in alphanumeric order for each routine. The mathematical notation is as defined in Volume I (ref. 1). The list for unlabelled common includes some variables which are equivalenced to variables in unlabelled common.

4.1 Unlabelled Common

AA(I,J)	Matrix of coefficients and constants for a system of simultaneous linear equations
AAA(I,J)	Equivalent to AA(I,J)
CDIJ(I,J)	Matrix of coefficients $\bar{v}_{i-c,j}$ for expressing the dependent species in terms of the c independent species
CAPX(J)	Mole fraction X_J for the J^{th} species
GJ(J)	Species concentrations in moles per gram of mixture; equivalenced to CAPX(J)
A	α , constant in the density-area relation I(383)
AFNTS	Effective area ratio calculated for equilibrium flow in the nonequilibrium solution by the perturbation method
AFNX	Effective area ratio, A_e
AMACH	Mach number

AR	Indicator (initially 0) which is set to 1 when $d \ln \tilde{A}_e / dx$ becomes positive downstream of the throat in the nonequilibrium solution by the inverse method.
ARBA	Maximum allowable number of tries at switching from upstream to downstream region in nonequilibrium solution
ARBB	Counter for number of times upstream-downstream switching point is moved downstream
BZERØ	b_0/W^0 , where b_0 is the covolume of the molecules in the high-density modification of the equation of state and W^0 is the molecular weight of the undissociated gas.
C	Constant C in density-area curvefit relation I(383)
CARB	Effective area ratio at the beginning of the current integration step
CH	Nondimensional specific enthalpy, hW_0/R_0T_0 , where h = specific enthalpy, W_0 = reservoir molecular weight, R_0 = universal gas constant, T_0 = reservoir temperature
CHA	Nondimensional specific enthalpy in the reservoir
CLNT	$\ln T$, where T = temperature in $^{\circ}K$
CM	Molecular weight of gas mixture, W (g/mole)
CMA	Molecular weight in reservoir, W_0 (g/mole)
CRA	R_0 , universal gas constant (1.9872 cal/mole $^{\circ}K$)
CRP	R_0T_0 , where T_0 = reservoir temperature ($^{\circ}K$)
CRRB	Intermediate variable in calculation of entropy

CRS	Dummy variable for entropy
CSTA	$0.5 \ln (W_0/R_0 T_0)$, where W_0 is the reservoir molecular weight (g/mole), R_0 the universal gas constant, and T_0 the reservoir temperature ($^{\circ}\text{K}$)
CT	Ratio of the temperature to the reservoir temperature, T/T_0
CTAP	Reservoir temperature, T_0 ($^{\circ}\text{K}$)
CTB	Dummy variable for CT
CTC	Dummy variable for CT
CTMAX	Ratio of throat temperature to reservoir temperature
CTMXX	Temperature for switching from thermo-fit to physical model for species properties, divided by reservoir temperature
CTP	Temperature in $^{\circ}\text{K}$
CTPL	$\ln T_0$
CTT	CT at the last printed step of the non-equilibrium solution
CX	Streamwise coordinate $x(\text{cm})$, zero at the throat and positive downstream
CXB	CX at the beginning of the current integration step
CXMAX	Maximum allowable value of $x(\text{cm})$
DATEST	Value of effective area ratio at which nonequilibrium solution is switched from upstream to downstream region
DBTEST	(Not used)

DELT1	Nondimensional temperature decrement used in frozen and equilibrium solutions
DELT2	(Not used)
DELTAX	Increment in x (cm)
DLØGA	$d \ln A_e / dx$
DLØGR	$d \ln (\rho / \rho_0) / dx$, logarithmic derivative of nondimensional density
DT	$d(T/T_0) / dx$, gradient of nondimensional temperature
ENT	Intermediate variable used in frozen flow solution
FLUX	Nondimensional mass flux, $(\rho / \rho_0) \cdot u / \sqrt{R_0 T_0 / W_0}$
HDELX	DELTAX/2
PCT	$\delta (T/T_0)$, perturbation in nondimensional temperature
PCTEST	Tolerance on C_χ when testing size of $\delta \chi_i$
PRES	Nondimensional pressure, p/p_0
PRESA	Reservoir pressure p_0 (atm)
PRESB	Nondimensional pressure at the preceding flow point; used in Mach number calculation
PRESTH	Nondimensional pressure at throat
PRHØ	Perturbation in nondimensional density, $\delta \rho / \rho_0$
RHAP	Density in reservoir, ρ_0 (g/cm ³)

RHØ	Nondimensional density, ρ/ρ_0
RHØB	RHØ at the preceding flow point; used in Mach number calculation
RHØBAR	Effective density in high-density equation of state divided by effective reservoir density
RHØC	(Not used)
RHØP	Density ρ in gm/cm ³
RHPL	$\ln \rho$
RHTH	Nondimensional density at throat, ρ_*/ρ_0
RØBARA	$\tilde{\rho}_0/\rho_0$
RØBARP	Effective density in reservoir, $\tilde{\rho}_0$ (gm/cm ³)
SCPG	$\sum_j \gamma_j c_{pj}/R_0$
SDT	Change in nondimensional temperature in integration interval
SEN	Entropy in cal/g °K
SHPG	$\sum_j \gamma_j H_j/R_0 T_0$
SC	Factor by which integration interval is increased
SL	Characteristic length, l , now set at 1 cm
SL64	$1/\sqrt{F}$, where F is the conversion factor from calories to ergs
SM	Nondimensional mass flux at the throat, $RHTH \cdot (SU)_*$
SU	Nondimensional velocity, $u \sqrt{W_0/R_0 T_0}$

SU2	$(SU)^2$
SUMG	$\sum \gamma_j$
TEST	Convergence criterion value for Newton-Raphson iterations in calculations of thermochemical equilibrium
TESTB	Tolerance allowed in locating temperature at throat for frozen and equilibrium flow
TPRINT	Nondimensional temperature interval at which results of nonequilibrium calculation are to be printed
TSTØP	Minimum value of nondimensional temperature desired in solution
UP	(Not used)
ZP	Natural log of nondimensional pressure
ZPA	Natural log of reservoir pressure in atm
BE(I)	$\beta_i = \sum_j \beta_{ij}$
BET(I)	$-1 + \nu^*_{i-c}$
BLBK(I)	Temporary storage area for boundary layer properties at the switch point from the upstream to downstream solutions
CAI(I)	A_i , constant factor in reaction rate formula
CAPQ(I)	Number of gram atoms of the I th chemical element per mole of the cold gas; equivalenced to BLBK(I)
CAPXTH(J)	X_{j*} , mole fraction of J th species at throat
CCI(J)	Temporary storage used in calculation of species molecular weights; equivalenced to CAPXTH(J)

CCPJ(J)	Molar heat capacity of the J th species divided by R ₀
CEACT(I)	Activation energy for i th reaction (cal/mole °K)
CGI(I)	Species molecular weights (g/mole)
CGMU(I)	$\ln X_i$, logarithms of mole fractions
CHI(I)	χ_i , variable measuring the departure of the i th reaction from equilibrium, eq. I(289)
CHII(I)	Intermediate variables in computation of equilibrium constant based on mole fractions
CLNIMC(I)	$\ln (1 - \chi_i)$
CLNPI(I)	$\ln P_i$, variables in nonequilibrium solution
CMW(I)	Atomic weights of elements (g/mole)
DGJ(J)	$d \gamma_j / dx$; equivalenced to GJA(J)
ETAI(I)	η_i , temperature exponent in rate constant for i th reaction
ETAJ(J)	n_j , number of atoms in a molecule of the J th species; 0 value indicates no physical model data for thermal properties
GJA(J)	γ_{j0} , concentration of j th species in reservoir
GJB(J)	GJ(J) at the beginning of the current integration step
PERTGJ(J)	$\delta \gamma_j$, perturbations in species concentrations
PGJ(J)	Intermediate variables in Newton-Raphson calculation of equilibrium composition.

PI(I)	Variables P_i in nonequilibrium solution, eq. I(288)
PICHI(I)	$P_i \chi_i$
QM(I)	Number of moles q_j of the j^{th} independent species in one mole of mixture
QQ(I)	1 if a third-body list is provided for the I^{th} reaction, 0 if not
SAJ(J)	$b_j + (1.5 + n_j) \ln T_0$, constant in physical-model expression for μ_j°
SBJ(J)	b_j , eqs. (51), Vol. I; equivalenced to SAJ(J)
SDCHI(I)	$\delta \chi_i$, perturbation in χ_i
SDGJ(J)	Change in γ_j over an integration step
SENT(J)	Nondimensional species entropy, S_j°/R_0
SHJ(J)	Nondimensional species enthalpy, $H_j/R_0 T_0$
SHJA(J)	Nondimensional formation enthalpy of j^{th} species, $H_{j0}^{\circ}/R_0 T_0$
SHJAP(J)	Formation enthalpy of J^{th} species, cal/mole
SKIL(I)	Intermediate variable in calculation of dependent species concentrations from concentrations of independent species
SS(J)	$\ln \gamma_{j0}$, logarithms of reservoir species concentrations
TB(J)	Temporary storage for quantities necessary to restart a step in the nonequilibrium calculation
TFA(J)	a_j , coefficient in thermo-fit expressions for H_j and μ_j°

TFB(J)	b_j , coefficient in thermo-fit expressions for H_j and μ_j^0
TFC(J)	c_j , coefficient in thermo-fit expressions for H_j and μ_j^0
TFD(J)	d_j , coefficient in thermo-fit expressions for H_j and μ_j^0
TFE(J)	e_j , coefficient in thermo-fit expressions for H_j and μ_j^0/T
TFK(J)	k_j , coefficient in thermo-fit expressions for H_j and μ_j^0
THEV(J)	Nondimensional characteristic vibrational temperature, θ_{vj}/T_0
THEVP(J)	Characteristic vibrational temperature for J^{th} species, θ_{vj} (°K)
XMJAT(J)	μ_j^0/R_0T
XNUI(I)	$\nu_i = \sum_j \nu_{ij}$, sum of stoichiometric coefficients on the reactant side of the i^{th} reaction
BETA(I,J)	$\beta_{ij} = \nu'_{ij} - \nu_{ij}$
BTA(I,J)	Equivalenced to BETA(I,J); not used
ELJ(L,J)	Nondimensional energy of L^{th} electronic level of J^{th} species, E_{lj}/R_0T_0
GELJ(L,J)	Degeneracy of L^{th} electronic level of J^{th} species, g_{lj}
XNUIJ(I,J)	ν_{ij} , stoichiometric coefficient of J^{th} species on reactant side of i^{th} reaction
XNUIJP(I,J)	ν'_{ij} , stoichiometric coefficient of j^{th} species on product side of i^{th} reaction
IC	Number of ions included in chemical model

IM	Index of the first species in the equilibrium flow calculation, incremented when electrons are dropped from calculation
INEQ	Indicator (initially 0), reset to 1 when numerical integration of nonequilibrium solution is begun
INEQV	Indicator for selecting whether equilibrium (0) or frozen (1) vibrational model is to be used
IP	(Not used)
IRUN	Run number for identification
ISC	Number of elements in mixture (including the electron, if ionized species are included)
ISCP1	ISC+1
ISMC	Number of species minus number of elements (n-c)
ISMCNR	Current value of (n-c) in equilibrium flow calculation (decreased by IC when electrons are dropped from the calculation)
ISR	r, number of reactions
ISS	n, number of species
ISSNR	Current value of n in equilibrium flow calculation (decreased by IC when electrons are dropped from calculation)
ISSP1	n+1
ISSP2	n+2
ISSP3	n+3
ISSP4	n+4

ISW1A	Input control variable; 0 suppresses frozen solution
ISW1B	Input control variable; nonzero value gives edits of transport cross section calculations
ISW2A	Input control variable; 0 suppresses non-equilibrium solution
ISW2B	Input control variable; selects types of input data for reservoir calculation
ISW3A	Input control variable; 0 suppresses equilibrium solution
ISW3B	Input control variable; 0 suppresses boundary layer calculations
ISW4A	Input control variable; must be nonzero if another case follows, 0 for last case
ISW4B	Input control variable; nonzero gives dump in boundary layer routine BLAYER
ISW5A	Input control variable; nonzero gives dump tracing the execution of subroutine RESTMP, which computes the reservoir temperature from mass flow and reservoir pressure or stagnation enthalpy
ISW5B	Input control variable, nonzero gives dump in subroutines EXACT, COMM, NONEQ and PRTA for debugging nonequilibrium calculations
ISW6A	Input control variable; if positive, only the reservoir equilibrium calculation is done; if negative, edit of species thermal properties is produced
ISW6B	Input control variable; 0 suppresses the output of species mole fractions in the free-stream and model-point output

IUPD	Indicator (initially 1), reset to 0 when switch from inverse method to direct integration is made in nonequilibrium solution
IZERØ	0 (set in READ)
JJK	Indicator (initially 0), reset to 1 in NEWRAP when electrons are dropped from the equilibrium calculation
LC	(Not used)
M1	Index equal to c+1
NFIT	Indicator. 0 if thermo fit data are not used for any species
NIT	Counter for number of iterations in Newton-Raphson procedure
NNN	Integration step counter
NNS	Number of times Δx has been increased by current value of SC
NQS	Number of successful integration steps before Δx is increased
NQT	(Not used)
NTEST	Number of iterations allowed in Newton-Raphson procedure
IGJ(J)	Indicator; 1 if thermo-fit data are used for J^{th} species, 0 if not
IGM(J)	Number of electronic levels for the J^{th} species
ITB(I)	Temporary storage for indices necessary to restart step in nonequilibrium calculation

KUR(I,J) U_{ij} , third body matrix
 LPIJ(I,J) α_{ij} , number of atoms of the J^{th} chemical element in a molecule of the I^{th} species
 ACØM(I) Hollerith description for case
 ELEMENT(I) Chemical symbols for elements (Hollerith)
 HP(I) Symbols for species (Hollerith)

4.2 Common /AEGEØM/

SQRTA Square root of the geometric area ratio
 S1 Geometric area ratio (reset to the effective area ratio in AESOLN)
 S2 Derivative of geometric area ratio (reset to derivative of the effective area ratio in AESOLN)

4.3 Common /AREA/

ATPI(J,I) Downstream boundary of the J^{th} section in the I^{th} profile (cm from the throat, positive downstream)*
 PARAM(L,J,I) For $L = 1$ to 3, the parameter values for the J^{th} section in the I^{th} profile. See Sec. 4.3 of Volume I
 RTHCM(I) Throat radius (cm) for the I^{th} profile
 NSECT(I) Number of sections in the I^{th} profile
 NSECTU(I) Number of upstream sections in the I^{th} profile
 ISHAPE(J,I) Shape index for J^{th} section in the I^{th} profile

*"ATPI" is an acronym for "area transfer point input"

NPROFL(I) Index of i^{th} profile in the precoded data; NPROFL(1) is equivalent to NOZZLE.

NPRFLS Number of profiles (1 for a nozzle, 2 for a channel)

NBL Index (1 or 2) of the profile which diverges from the channel axis least rapidly downstream of the throat

4.4 Common /AVG/

WSAVE Parameter controlling the averaging distance for the boundary layer correlation parameter, n

4.5 Common /BL/

DELBL(L) Displacement thickness of the boundary layer on the L^{th} profile divided by R_0 . For two-dimensional and axisymmetric nozzles, only DELBL(1) is used.

BLINT(L) Integral I for the boundary layer on the L^{th} profile, eq. I(172)

XZERØ Position at which boundary layer is assumed to begin (negative value, cm from the throat)

TWALL Nozzle wall temperature ($^{\circ}\text{K}$)

CPWALL Specific heat of the gas at the wall temperature (cal/g $^{\circ}\text{K}$)

VISRØT Ratio of viscosity in the reservoir to the reservoir temperature (poise/ $^{\circ}\text{K}$)

DIAM(L) Throat diameter for the L^{th} profile (inches)

SW $S_w = (h_w/h_0) - 1$, where h_w is the gas enthalpy at the wall and h_0 the free-stream stagnation enthalpy

R0 Characteristic length used in boundary layer calculations (cm); equal to the throat radius for two-dimensional and axisymmetric nozzles; equal to $\sqrt{A'_*/\pi}$ for channels, where A'_* is the geometric cross sectional area of the channel at the throat

JDIM Dimensionality index, 0 for two-dimensional nozzles, 1 for axisymmetric nozzles

IPØINT Index of flow solution points at which boundary layer calculations are done

4.6 Common /BLNE/

XI(I) For $i = 1$ and 2 , the streamwise coordinate along the nozzle or channel surface for the I^{th} profile (cm)

ØRDINP(I) The quantity δ for the boundary layer on the I^{th} profile at the preceding flow point, eq. I(171)

DELBLP(I) $\delta^*/R0$ for the boundary layer on the I^{th} profile at the preceding flow point

DDBLP(I) $(d\delta^*/dx)/R0$ for the boundary layer on the I^{th} profile at the preceding flow point

AMPP Mach number at the point preceding the previous flow point

AMP Mach number at the preceding flow point

CØUPLD Logical variable, set to .TRUE. when the coupling of the boundary layer to the inviscid flow is switched on

ISMD The quantity (30-i) in eq. I(418)

4.7 Common /BLOUT/

REPF Reynolds number per foot (ft^{-1})

THETA(L) Momentum thickness for the boundary layer on the L^{th} profile, divided by R_0

SN(L) Correlation parameter n , eqs. I(158) and I(174), for the boundary layer on the L^{th} profile

XSN(L) Averaged correlation parameter \bar{n} , eq. I(217), for the boundary layer on the L^{th} profile

PRREF Prandtl number at the reference temperature

HR Recovery enthalpy (cal/g)

QWDØT(L) Heat flux to the wall surface with the L^{th} profile ($\text{Btu}/\text{ft}^2\text{sec}$)

TAUW(L) Shear stress on the wall surface with the L^{th} profile ($\text{lb-f}/\text{ft}^2$)

4.8 Common /BLRAD/

YOZO Product of the two throat radii for a channel (cm^2)

4.9 Common /CERRØR/

FAILED Logical variable, set to .TRUE. when a convergence test on an integration step is failed in subroutine DERIVS or RNKT

4.10 Common /CHAN/

CPl(I) Specifications for standard channel no. 1:

I = 1 Index of first channel profile in compiled-in list of profiles

- I = 2 Index of second channel profile in compiled-in list of profiles
- I = 3 Hollerith name for channel
- I = 4 Index (1 or 2) of the profile which diverges from the channel axis least rapidly downstream of the throat
- I = 5 Hollerith name for facility

- CP2(I) Specifications for standard channel no. 2
- CP3(I) Specifications for standard channel no. 3
- CP4(I) Specifications for standard channel no. 4
- CP5(I) Specifications for standard channel no. 5

4.11 Common /COLDSP/

- CGMW Molecular weight of the cold gas mixture (g/mole)
- QPJ(J) Mole fraction of the Jth species in the cold gas mixture
- IJCS(J) Index of the Jth cold species in the list of species for the current problem
- NCS Number of cold species

4.12 Common /CONVRT/

- CF(1) Conversion factor from nondimensional temperature CT to temperature in °K
- CF(2) Conversion factor from nondimensional pressure PRES to pressure in atm
- CF(3) Conversion factor from nondimensional density RHØ to density in lbm/ft³

- CF(4) Conversion factor from nondimensional velocity SU to velocity in ft/sec
- CF(5-7) Conversion factors equal to 1
- CF(8) Conversion factor from viscosity in poise to viscosity in lbm/ft sec
- CF(9) Conversion factor equal to 1
- CF(10) Conversion factor from length in cm to length in inches
- CF(11) Conversion factor from nondimensional enthalpy CH to enthalpy in Btu/lbm
- CF(12) Conversion factor equal to 1
- CF(13-16) Conversion factors from nondimensional boundary layer thicknesses DELBL and THETA to thicknesses in inches
- CF(17-26) Conversion factors equal to 1

4.13 Common /DRV/

- SHJDGJ $(R_0 T_0)^{-1} \sum H_j d \gamma_j / dx$, where H_j is the molar enthalpy of the jth species and γ_j is the concentration of the jth species in moles per gram of mixture (set in subroutine EXACT)

4.14 Common /EELEM/

- EEPl(J) Data for first input-defined element
 J = 1 Atomic number
 J = 2 Atomic weight (g/mole)
- EEPl0(J) Data for tenth input-defined element
 J = 1 Atomic number
 J = 2 Atomic weight (g/mole)

4.15 Common /ELEM/

EP1(I) Specifications for standard chemical element
no. 1:

EP1(1) Hollerith name
EP1(2) Atomic weight (g/mole)

EP2(I) Specifications for standard chemical element
no. 2

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· ·
· ·

EP10(I) Specifications for standard chemical element
no. 10

4.16 Common /EQC/

This block is used by subroutine EQCALC to communicate
the results of its thermochemical equilibrium calculations
to the calling routine.

ZCAP(J) Mole fraction X_j of Jth species

ZSEN Entropy (cal/g °K)

ZCH Nondimensional molar enthalpy, $(R_0 T_0)^{-1} \sum X_j H_j$

ZCM Mean molecular weight

ZRBP Effective density ($\tilde{\rho}_0$) in reservoir for
imperfect-gas model

ZRHØ Nondimensional density, ρ/ρ_0

4.17 Common /EQC2/

ZPZ $\ln (p/p_0)$, logarithm of the nondimensional
pressure

ZGMU(I) $\ln X_i$, logarithm of the mole fraction for the
Ith chemical species

4.18 Common /ERROR/

ERR Logical flag set to "true" in DUMP routine; if "true", control is passed back to Main program, DUMPEX is called to print a dump of common data, and case is terminated

4.19 Common /GLIM2/

NØREAC(I) Logical control variables; when NØREAC(I) is set to .TRUE., the Ith reaction is suppressed

4.20 Common /INGNE/

AIN(K,I) The inverse A_{ki} of the square submatrix α_{ij} of the matrix α_{ij} , specifying the number of atoms of the jth element per molecule of the ith species; see eq. I(3)

4.21 Common /LN/

ISATØM Index of atom used in Lewis number calculation, in master list of species

ISMØL Index of molecule used in Lewis number calculation, in master list of species

JATØM Index of atom used in Lewis number calculation, in list of species for current problem

JMØL Index of molecule used in Lewis number calculation, in list of species for current problem

4.22 Common /MASSFL/

SMASS Sonic mass flux based on total mass flow and geometric throat area (g/cm²sec)

CTMXXI Temperature (°K) above which species thermal properties are computed from the thermo fit for those species for which thermo fits are supplied

TSTØPI Free-stream temperature at which the flow solutions will be terminated (^oK)

IS(J) Index of the Jth species in the current problem, in the master list of species

4.23 Common /MIXT/

GPI(I) Specifications of standard gas mixture no. 1:

- I = 1 Hollerith name of mixture
- I = 2 Number of chemical elements in mixture, including e⁻ if the gas model contains ion species
- I = 3 Number of chemical species in mixture, including e⁻ if model contains ion species
- I = 4 Number of reactions included in gas model
- I = 5 Number of ions in gas model, excluding e⁻
- I = 6-15 Indices of elements present in mixture, in master list of elements; if electrons are present, they are the first element
- I = 16-25 Mole fractions (QPJ) of the cold species
- I = 26-45 Indices of the species included in the gas model, in the master list of species
- I = 46-109 Indices of the reactions included in the model, in the master list of reactions

I = 110-119	Indices of the cold species, in the master list of species
I = 120	Number of cold species
I = 121	Atom index for Lewis number calculations, in the master list of species
I = 122	Molecule index for Lewis number calculations, in the master list of species
I = 123	INT. If INT = 0, electron temperature equals gas temperature. If INT > 0, the model includes electronic nonequilibrium, and INT is the index of the extra reaction properties required in TNEP(I,INT)
I = 124	Indicator for inclusion (1) or exclusion (2) of Fay-Riddell Lewis number factor in the stagnation-point heat flux
GP2(I)	Specifications of standard gas mixture no. 2
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.	.
.	.
GP6(I)	Specifications of standard gas mixture no. 6

4.24 Common /MODPAR/

XMP1	Initial distance from throat at which model condition calculations are to be done (cm)
DXMP	(Not used)
FSTAG	Control variable; 0. value suppresses frozen shock calculations at model points, negative value suppresses equilibrium shock

CATFAC	Catalytic efficiency parameter $\bar{\alpha}$ for stagnation point heat flux calculations; 0 for noncatalytic wall
TMØDEL	Model wall temperature for stagnation point heat flux calculations ($^{\circ}$ K)
XMØDPI	Initial distance from the throat at which model condition calculations are to be done (inches)
DXMØDP	(Not used)
TPLATE	Wall temperature for calculations of heat flux to a flat plate at zero angle of attack ($^{\circ}$ K)
KDIM	Control variable for stagnation point heat flux calculations; 0 for two-dimensional model geometry, 1 for axisymmetric model

4.25 Common /MØDPT/

TSDIAM(I)	Test-section diameters at which model condition calculations are to be done (inches); in channel flow problems, specified channel widths at which flow calculations are to be done
TSAR(I)	Geometric area ratios at the nozzle stations defined by TSDIAM(I)
NTS	Number of specified test section diameters
MBL	In channel flow problems, index of the profile defining the channel width

4.26 Common /MØDP2/

MØDLPT	Logical indicator, .TRUE. when the current step in the nonequilibrium solution is at a model point
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4.27 Common /NEQ/

ØMDST(L) 1 - δ^*/R_0 for the Lth profile at the sonic point

DDELBL(L) (d δ^*/dx)/ R_0 , derivative of nondimensional displacement thickness for the boundary layer on the Lth profile

4.28 Common /NEWMP/

FACMP Factor by which distance from throat is increased in a geometric sequence of model points

NMØDPT Number of model points in geometric sequence

4.29 Common /NØZZ/

ZP1(i) Specifications of standard nozzle profile no. 1:

I = 1 Throat radius in cm

I = 2 Distance upstream of the throat of the point at which the boundary layer is assumed to originate (negative value, cm)

I = 3 Number of profile sections upstream of the throat

I = 4 Number of profile sections downstream of the throat

I = 5-16 Shape indices (ISHAPE) of the profile sections (in order from upstream to downstream):

 ISHAPE = 1 Straight section

 ISHAPE = 2 Circular section convex toward axis

ISHAPE = 3 Circular section concave
toward axis

I = 17-27 Downstream boundaries of profile
sections (distances from the throat,
positive downstream, cm)

I = 28-63 Parameters defining the profile
sections (lengths in cm).* There
are three parameter values P_1 , P_2 ,
 P_3 for each profile. For ISHAPE =
1, the equation of a straight
profile is

$$r(x) = P_1 + P_2x$$

For ISHAPE = 2 or 3

P_1 = distance of circle center
from axis

P_2 = x-coordinate of circle center

P_3 = circle radius

I = 64 Hollerith facility name

ZP2(I) Specifications of standard profile no. 2

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ZP20(I) Specifications of standard profile no. 20

4.30 Common /ØUIMØD/

TB7 $N_0 \rho_0$, where N_0 is Avogadro's number (6.0225×10^{23} particles per mole) and ρ_0 is the density in the reservoir in g/cm^3 .

*See Section 4.7 of Volume II

4.31 Common /ØUTPUT/

FVØUT(I) Array used for output of flow variables; set
in subroutine ØUT1 (entry ØUT2)

- I = 1 Axial coordinate, x (inches)
- I = 2 Temperature ($^{\circ}$ K)
- I = 3 Static enthalpy (Btu/lb)
- I = 4 Pressure (atm)
- I = 5 Density (lb/ft³)
- I = 6 Nozzle diameter (inches); geometric
area ratio in channel solutions
- I = 7 Velocity (ft/sec)
- I = 8 Mach number
- I = 9 Entropy (Btu/lb^ØR)
- I = 10 Frozen specific heat ratio
- I = 11 Effective area ratio
- I = 12 Reynolds number (per ft)
- I = 13 Molecular weight (g/mole)
- I = 14 Viscosity (lb/ft sec)
- I = 15 Electrical conductivity (mho/cm)
- I = 16 Geometric area ratio; width of face
with first profile in channel
solutions
- I = 17 Displacement thickness (inches)
- I = 18 Momentum thickness (inches)

- I = 19 Heat flux to wall (Btu/ft²sec)
- I = 20 Shear stress on wall (lbf/ft²)
- I = 21 Recovery enthalpy (Btu/lb)
- I = 22 Prandtl number at the reference temperature
- I = 23 Stanton number
- I = 24 Reynolds number based on θ
- I = 25 Re_{θ} for boundary layer transition
- I = 26 Width of face with second profile (channel solutions only)
- I = 27 Displacement thickness on second profile (inches)
- I = 28 Momentum thickness on second profile (inches)
- I = 29 Heat flux to wall on second profile (Btu/ft²sec)
- I = 30 Shear stress on wall on second profile (lbf/ft²)
- I = 31 Recovery enthalpy (Btu/lb)
- I = 32 Prandtl number at the reference temperature
- I = 33 Stanton number on second profile
- I = 34 Re_{θ} on second profile
- I = 35 Re_{θ} for transition

GJMF(J)

(Not used)

4.32 Common /POLYAT/

THEVE(K,J) For K = 1-4, the four characteristic vibrational temperatures of the Jth species in the current gas model, if this species is a linear triatomic molecule. In subroutine READ, the entries K = 2-4 are set to values in °K. In subroutine INIT, these values are nondimensionalized by dividing them by the reservoir temperature, and the value K = 1 is set

4.33 Common /RDLIST/

IGAS Index of standard gas mixture

IGASØ IGAS value in preceding case (if any)

NØZZØ Nozzle profile index for preceding case

ICHANØ Channel index for preceding case

4.34 Common /RDMAIN/

HS Input value of stagnation enthalpy (cal/g)

SUPGØ Logical control for suppressing output of the gas model data when the model is the same as in the preceding case

MFITER Input control for the iteration to take the displacement thickness into account in the reservoir condition calculations; zero value suppresses the iteration

NØ^mRAN Input logical control for suppressing all transport property calculations

4.35 Common /RDMØD/

LEWIS Index controlling use of the Fay-Riddell Lewis number factor in stagnation-point heat flux calculations. For LEWIS = 1, the factor is used; for LEWIS = 2, it is omitted

IAMBIP Index specifying whether the diffusion process upon which the Lewis number is based is ambipolar (IAMBIP = 2) or not (IAMBIP = 1)

4.36 Common /RDØUT/

FLØW Total mass flow in lbm/sec

FACNAM Hollerith facility name

CHANAM Hollerith channel name

LIMØUT Index, nonzero if the boundary layer parameters N and XSN are to be printed in the output, equal to 0 if not

4.37 Common /RDTR/

ISW8B Index controlling diagnostic dumps in the transport property routines. If ISW8B = 0, these dumps are omitted. If ISW8B > 0, the PUTQIN dump is produced once every ISW8B times the subroutine PUTQIN is called. If ISW8B < 0, the PUTQIN dump is suppressed.

4.38 Common /RDWEDG/

ANGLE(I) Angles of attack (degrees)

RADLE(J) Radii of leading edge (inches)

WX1 Initial distance from leading edge of wedge (inches)

DWX Increment in distance from leading edge (inches)

WXI(K)	Specified distances from leading edge (inches)
TWEDGE	Surface temperature of wedge model ($^{\circ}$ K)
WK	Leading-edge drag coefficient, k
NWX	Number of distances from leading edge (at increment of DWX) at which surface conditions on wedge model are to be calculated
NANGLE	Number of angles of attack
NRADLE	Number of leading-edge radii
WEDGEM	Logical control variable; if "false", conditions on wedge are not calculated
AXISYM	Logical control variable; if "false", stagnation point heat fluxes are not calculated
ISW9B	Control variable, normally 0, with the following possible optional values: ± 1 Print shock ordinate Y_s ± 2 Print nondimensional coordinate ζ ± 3 Print both Y_s and ζ If ISW9B is negative, calculations are done using the unmodified Cheng-Kemp theory as well as the modified theory

4.39 Common /REAC/

RPl(I)	Specifications of standard reaction no. 1:
I = 1	Coefficient A in formula for forward reaction rate constant (cm ³ /mole sec or cm ⁶ /mole ² sec)
I = 2	Exponent η in formula for rate constant

TPRNTI	The free-stream nonequilibrium solution is printed out at temperature intervals greater than or equal to TPRNTI times the reservoir temperature. For TPRNTI = 0., every step is printed
DELTXI	Input value of initial step size in x for nonequilibrium integration (cm)
CXMAXI	Distance beyond the throat at which the flow solutions are stopped (inches)
CTAPI	Reservoir temperature input ($^{\circ}$ K)
XZERØI	Nozzle or channel inlet position at which the boundary layer is assumed to originate (negative value, measured in inches upstream from the throat)
PARAMI (L,M,N)	Input array of parameter values for nozzle or channel profiles (see definitions of code inputs, Group 4, in Section 2.3 of Volume II)
BZERØI	Input of constant in imperfect gas correction
HSTAG	Input stagnation enthalpy (Btu/lb)
READXS	Input control for reading cross section data for transport property calculations
READG	Input control for reading data on elements, species, and/or reactions
AAMS	Input control for automatic air model selection
AXIMØD	Input control for suppressing stagnation point model condition calculations if only wedge calculations are desired
ICASE	Case number in the current job
NEELS	Number of elements being defined in the input

ISW7B	Input control for suppressing the output of the boundary layer parameters N and XSN
INT	Indicator for electronic nonequilibrium in a gas model
ICHAN	Index of standard channels
NQSI	Number of successful integration steps required before the step size in the non-equilibrium calculation is increased
INEQVI	Input control for equilibrium or frozen molecular vibration
NRECO	Input of number of records already on data tape at beginning of run
JCS(I)	Indices of cold species in master list of species
ISCI	Input of the number of chemical elements in the gas model
ISSI	Input of the number of chemical species in the gas model
ISRI	Input of the number of reactions in the gas model
ICI	Input of the number of ions in the gas model
IE(I)	Input indices of elements present in the gas model, in the master list of elements
IR(I)	Input indices of reactions in the gas model, in the master list of reactions
NSECTS(I,J)	For I = 1, number of upstream sections in the curvefit for the Jth profile; for I = 2, number of downstream sections
IEEP(I)	Indices assigned to the input-defined elements in the master list of elements

4.41 Common /RENE/

GAMIN	Concentration (moles/g) below which a species will be frozen if it decreases so rapidly that it controls the integration step size.
HTEST	Maximum relative change in total enthalpy in an integration step.
TETEST	Maximum relative change in electron temperature in an integration step
QTEST	Criterion value for maximum allowable change in energy transfer to electron gas in an integration step
DCHLL	Parameter used in selecting initial integration step size; the initial Δx is not allowed to exceed $0.01 \delta \chi_i _{\min} / DCHLL$.
DCHRAT	Parameter controlling the artificial increase in reaction rates in the perturbation solution to avoid premature startup of the numerical integration (Section 7.3.7 of Volume I)
CCHI	Parameter C_χ in criterion for switch from perturbation solution to numerical integration
TTEST	Maximum relative change in the gas temperature in an integration step
GTEST	Maximum relative change in a species concentration in an integration step.

4.42 Common /RESRP/

VISCR	Viscosity in the reservoir (poise)
PRR	Prandtl number in the reservoir
SIGR	Electrical conductivity in the reservoir (mho/cm)

FLEWR Lewis number in the reservoir

4.43 Common /SPEC/

SPl(I) Specifications for standard species no. 1:

- I = 1 Hollerith name for species
- I = 2 Number of elements in species (≤ 3)
- I = 3-5 Indices of elements in the master list of elements
- I = 6-8 Numbers of atoms of elements
- I = 9-14 Thermo-fit parameters a, b, c, d, e, k
- I = 15 Formation enthalpy (cal/mole)
- I = 16 Number of atoms in a molecule of the species
- I = 17 Chemical constant, b
- I = 18 Characteristic vibrational temperature ($^{\circ}\text{K}$)
- I = 19 Number of electronic levels (≤ 10)
- I = 20 1 if thermo-fit is used, 0 if not
- I = 21-30 Degeneracies of electronic levels
- I = 31-40 Energies of electronic levels (cal/mole above ground state)
- I = 41-43 Second, third, and fourth characteristic vibrational temperatures for linear triatomic species ($^{\circ}\text{K}$)

4.44 Common /SS/

- CAS Square of the sound speed divided by the non-dimensional temperature T/T_0 (cm^2/sec^2)
- US Standard velocity used for nondimensionalizing the flow velocity, $\sqrt{R_0 T_0 / W_0}$, where T_0 is the reservoir temperature, W_0 the reservoir molecular weight, and $R_0 = 8.31436 \times 10^7$ erg/mole $^\circ\text{K}$

4.45 Common /STAG/

- T2E Initial estimate of temperature behind equilibrium normal shock ($^\circ\text{K}$)
- EPSLØN Initial estimate of density ratio across equilibrium normal shock
- T2F Initial estimate of temperature behind frozen normal shock ($^\circ\text{K}$)
- EPSF Initial estimate of density ratio across frozen normal shock

4.46 Common /STEPS/

- NSTEPS Number of successful integration steps since the last printout of conditions at a flow point

4.47 Common /SWITCH/

- XPB The corrected x-coordinate of the flow point preceding the point at which BLAYER is first called with x corrected for the displacement thickness in the frozen and equilibrium solutions (cm)
- DXPB The difference in corrected x values between the point at XPB and the preceding point

W Constant parameter w determining the characteristic distance over which the boundary layer correlation parameter n is averaged; see eqs. I(218) and I(217)

4.48 Common /TAPØUT/

XXX(I) Block of storage used to set up data which are to be saved on binary tape for subsequent processing by other programs

ITPØUT File number of binary output tape

NRCØUT Number of records written on binary tape during current case of job

IFLØW (Not used)

ITYPER Type of record written on binary tape:

- 1 "Case" record with overall definition of flow solution being computed
- 2 "Point" record containing free-stream flow variables
- 3 "Model" record containing data for stagnation point model

IMP Index of model points written onto binary tape in current solution

DATAPE Logical control variable; if "false", no data are written onto binary tape

4.49 Common /TEMPRY/

SAVEC(J) Species mole fractions for which transport property calculations are done

4.50 Common /THRT/

RSA Factor by which cross sectional area of flow has been rescaled in nonequilibrium solution

4.51 Common /TNCE/

- SUMGH $\sum_{j=2}^n \gamma_j$, sum of the molar concentrations for all species except the electron (mole/g)
- SCPGH $\sum_{j=2}^n C_{pj} \gamma_j / R_0$, sum of the products of the molar heat capacities and molar concentrations for all species except the electron, divided by the universal gas constant
- QDPR Power radiated from the gas, cal/cm³-sec
- QDPE Net power transferred to the electron gas (cal/cm³-sec)

4.52 Common /TNDC/

- XMJATD (J,L) Chemical potential at standard pressure (μ_j^0), for the Jth species, evaluated at the Lth temperature (L = 1, heavy particle temperature; L = 2, electron temperature), divided by R_0 times the Lth temperature
- CLNTD (L) Natural logarithm of the Lth temperature (L = 1, heavy particle temperature; L = 2, electron temperature)
- CTD (2) Nondimensional temperatures (L = 1, T/T_0 ; L = 2, T_e/T_0)

4.53 Common /TNE/

- TN1 (I) Specifications of data for standard electronic nonequilibrium model no. 1 (helium):
- I = 1-25 KTF (I), indicator for type of formula for forward rate constant of the Ith reaction; see Sec. 4.4 of Vol. II

- I = 26-50 KTR indicators for the reverse reactions in the gas model. KTR = 0 if the backward rate constant k_r is zero; KTR = 1 if $k_r = k_r(T)$; and KTR = 2 if $k_r = k_r(T_e)$
- I = 51-75 ITR values for the reactions in the gas model. ITR is an indicator of the rule for partitioning the reaction energy between the electrons, the heavy particles, and radiative losses. The significance of its values is discussed in Sec. 4.4 of Vol. II
- I = 76-100 ϵ_0 parameters for reactions
- I = 101-125 Values of the parameter "a" for reactions with ITR = 1
- I = 126-155 Temperature values for table of elastic collision cross section, $\bar{Q}(1,1)$
- I = 156-185 $\bar{Q}(1,1)$ values for table
- I = 186 Parameter b for reactions with KTF = 4

TN2(I) Specifications of data for electronic non-equilibrium model no. 2 (argon)

4.54 Common /TNEQ/

TLIST(1) For I = 1-30, temperatures for elastic collision cross section table for current gas model ($^{\circ}\text{K}$)

PØM(I) For I = 1-30, cross section values for elastic collision cross section table for current gas model

4.55 Common /TNERK/

SDTE	$\Delta T_e/T_0$, change in electron temperature in integration step
CTEB	T_e/T_0 at the start of the integration step
DCHA	$(W_0/R_0T_0) (dh_0/dx)$, rate of change of non-dimensional total enthalpy
CHB	W_0h_0/R_0T_0 , nondimensional total enthalpy at start of integration step
SDCHA	$\Delta (W_0h_0/R_0T_0)$, change in nondimensional total enthalpy in integration step
DQMAX	D_{qm} , criterion value for convergence test on changes in energy transfer to the electron gas
IFAIL	Indicator for cause of integration step failure:

<u>IFAIL</u>	<u>Meaning</u>
0	Failure in COMM or DERIVS (called from RNKT)
1	Test on T in NØNEQ
2	Test on T_e in NØNEQ
3	Test on h_0 in NØNEQ
4	Test on DL_{JGA} in NØNEQ
5	Test on QDPE in NØNEQ
11-30	Test on GJ(IFAIL-10) from RNKT in NØNEQ
31-50	Test on GJ(IFAIL-30) from element conservation in NØNEQ

<u>IFAIL</u>	<u>Meaning</u>
60	Failure in DERIVS or COMM (called from NØNEQ)
-1 to -ISS	Test on sign of GJ(-IFAIL) in RNKT (GJ2 or GJ3)
-(ISS+1)	Test on sign of T in RNKT (GJ2 or GJ3)
-(ISS+2)	Test on sign of T_e in RNKT (GJ2 or GJ3)
-(ISS+3)	Test on sign of h_0 in RNKT (GJ2 or GJ3)
-(ISS+4)	Test on QDPE in RNKT (GJ2 or GJ3)
-31 to (-30+ISS)	Test on sign of GJ(-IFAIL-30) in RNKT (GJ4, Runge-Kutta)
-(31+ISS)	Test on sign of T in RNKT (GJ4, Runge-Kutta)
-(32+ISS)	Test on sign of T_e in RNKT (GJ4, Runge-Kutta)
-(33+ISS)	Test on sign of h_0 in RNKT (GJ4, Runge-Kutta)
-(34+ISS)	Test on QDPE in RNKT (GJ4, Runge-Kutta)
-61 to -(60+ISS)	Test on sign of GJ(-IFAIL-60) in RNKT (GJ4, Treanor)
-(61+ISS)	Test on sign of T in RNKT (GJ4, Treanor)
-(62+ISS)	Test on sign of T_e in RNKT (GJ4, Treanor)

IFAILMeaning

- (63+ISS) Test on sign of h_0 in RNKT (GJ4, Treanor)
- (64+ISS) Test on QDPE in RNKT (GJ4, Treanor)

4.56 Common /TNONEQ/

- CTE Electron temperature divided by gas temperature in reservoir
- DTE Derivative of CTE with respect to x , i.e., $d(T_e/T_0)/dx$
- BPAR Parameter b for reactions with $KTF = 4$
- EPAR(I, JR) EPAR(1, JR) = parameter ϵ_0 for the JRth reaction in cal per N_0 reactions; EPAR(2, JR) = parameter a for the JRth reaction if ITR(JR) = 1
- NT Indicator for electronic nonequilibrium in the gas model; for $NT = 1$, the gas and electron temperatures are assumed to be equal; for $NT = 2$, they are not
- ITR(JR) Indicator of the rule for partitioning the reaction energy between the electrons, the heavy particles, and radiative losses in the JRth reaction of the current gas model. See discussion in Section 4.4 of Vol. II
- KTF(JR) Indicator for temperature dependence of the forward rate constant k_f for the JRth reaction in the current gas model; see Sec. 4.4 of Vol. II

KTR(JR) Indicator for temperature dependence of the backward rate constant k_r for the JRth reaction in the current gas model:

$KTR = 0$	$k_r = 0$
$KTR = 1$	$k_r = k_r(T)$
$KTR = 2$	$k_r = k_r(T_e)$

ICH(J) Indicator for charged species; equal to 1 if the Jth species is neutral, equal to 2 if the species is an ion

IPA(I) Species index for the product atom in the Ith reaction; if $KTF(I)$ is not equal to 4, $IPA(I)$ is set to zero

4.57 Common /TRANI/

III(K) For $K = 1$ to a maximum of 5, the index of the first species in the Kth pair of species to which the cross section calculated in the i^{th} step of the transport property calculation is applied. The common block provides storage for $i = 1$ to 100

4.58 Common /TRANJ/

JJi(K) For $K = 1$ to a maximum of 5, the index of the second species in the Kth pair of species to which the cross section calculated in the i^{th} step of the transport property calculation is applied. The common block provides storage for $i = 1$ to 100. Only pairs with $III(K) \leq JJi(K)$ are used

4.59 Common /TRANS1/

T Temperature in $^{\circ}K$ for transport calculations

Q(K,I,J)

Q(K,I,J) is used in the cross section computations to store the averaged cross sections for the species pair (I,J) as follows:

$$Q(1,I,J) = \bar{\Omega}_{ij}(1,1),$$

$$Q(2,I,J) = \bar{\Omega}_{ij}(2,2),$$

$$Q(3,I,J) = B_{ij}^* \bar{\Omega}_{ij}(1,1).$$

At other stages in the calculations, Q(K,I,J) is also used in subroutine TRANSP to store the matrix elements for the transport calculations, and in subroutine XSECT to indicate the species pairs for which cross section data have been supplied

ZM2(I)

Quantity proportional to molecular weight of species I, eqs. (171)

4.60 Common /TRANS2/

KKQ(L)

Index of option to be used for step L of cross section computations in input or precoded data; see Section 4.4 of Vol. II

NNQ(L)

Number of species pairs in IIQ, JJQ arrays for which option KKQ(L) is specified in input data

ISEQ(K)

Sequencing array specifying the order in which the defined steps of the transport cross section calculation are to be carried out

NNKQ

Number of steps specified for cross section computations

I(J)

Temporary array of species indices used by subroutine XSECT in preparing edited cross section data

4.61 Common /TRANS3/

B(I,L) Intermediate variable in transport calculations;
see KINT and KANDMU, Sections 3.27 and 3.28

BR(I,L) Reciprocal of B(I,L)

A(L) Intermediate variable in transport calculations;
see KANDMU, Section 3.27

X(I) Mole fraction of species I in gas mixture

DH(I) Nondimensional internal specific heat
($W_i C_{pi} / N_0 k - 5/2$) for species I in gas mixture

4.62 Common /TRANS4/

TL(I) Independent variable (temperature) for tabulated
cross section data

ØMEGAL(I) Value of first parameter determining cross
sections at TL(I) for tabulated cross section
data

ASTAR(I) Value of second parameter determining cross
sections at TL(I) for tabulated cross section
data

BSTAR(I) Value of third parameter determining cross
sections at TL(I) for tabulated cross section
data

4.63 Common /TRANS5/

N Number of species included in transport
calculations

IELEC IELEC=0 indicates that electrons are not
included in the calculations

ID1 Index in the current gas model of first
species used in Lewis number computation

ID2 Index in the current gas model of second species used in Lewis number computation

4.64 Common /TRANS7/

V(M) List of parameters for cross section computations in edited data

KQ(L) Option to be used for step L of cross section computations in edited data

NQ(L) Index of last species pair in IQ,JQ arrays to which step L of the edited cross section computations is to be applied

IQ(N) List of first indices for species pairs to which edited cross section data are to be applied

JQ(N) List of second indices for species pairs to which edited cross section data are to be applied

NKQ Number of steps in cross section computations for edited cross section data

4.65 Common /TRANS8/

NV(K) Number of parameters V required for option K of the cross section computations

N Number of species included in cross section computations

4.66 Common /TRANV/

VVi(K) For K = 1 to a maximum of 5, the input parameters for the *i*th step of the cross section calculations. The common block provides storage for *i* = 1 to 100

4.67 Common /TRPRØP/

VISC Viscosity of mixture in poise
PRF Frozen Prandtl number for mixture (dimensionless)
SIGMA Electrical conductivity of mixture in mhos/cm
FLEWIS Atom-molecule Lewis number of mixture (dimensionless)

4.68 Common /WMØD/

TSF Frozen stagnation temperature (°K)

4.69 Main Program

ATEST Previous value of A in the Newton-Raphson iteration to determine the parameter α in the density-area relation I(383)
ATH Effective cross-sectional area of the flow at the throat, with allowance for the boundary layer displacement thickness (cm²)
CHTH Nondimensional enthalpy hW_0/R_0T_0 at the throat
CMTH Molecular weight of gas mixture at the throat
D $(\rho_*/\rho_0)^A (A+2) - 2$, where ρ_* is the density at the sonic point in the equilibrium flow solution and A is the current estimate of the parameter α in the density-area relation
DELT Initial decrement of the nondimensional temperature for starting the perturbation calculation of the nonequilibrium flow
D1 $(\rho_*/\rho_0)^A [(A+2) \ln (\rho_*/\rho_0) + 1]$
ELTMIN Case execution time in minutes

ET	Elapsed time from beginning of execution (seconds)
ETO	Elapsed time from beginning of execution at the start of a case (seconds)
GJ(I)	Species concentrations in moles/g
GJA(I)	Species concentrations in the upstream reservoir, moles/g
GMF(J)	Mole fraction of the J th species under equilibrium sonic conditions
HOP	Reservoir enthalpy (cal/g)
I	D \emptyset index; also, index, in the list of species for the current problem, of the K th cold species
IRC \emptyset	Indicator; value 1 indicates that call to TRANSX was skipped in the previous case because tables of species thermal properties were generated
IS \emptyset LN	Indicator for type of flow solution: IS \emptyset LN = 1 Frozen IS \emptyset LN = 2 Equilibrium IS \emptyset LN = 3 Nonequilibrium
IS3	Storage for input value of ISW3B; used to restore value after a case in which ISW3B was set to zero because N \emptyset TRAN = .TRUE.
K	Cold species index
N	Iteration counter in Newton-Raphson solution for parameter α in area-density relation
NCASE	Counter for cases in a run
NC \emptyset MPL	Number of cases successfully completed in a run

NERRØR	Number of failed cases in a run
NØZZLE	Nozzle profile index
NRCTØT	Total number of records written on tape 8 during a run
ØNE	1.0
PIV	3.14159
SUM1	$\ln (\rho_*/\rho_0)$
SUM2	A+2
SUM3	$(\rho_*/\rho_0)^A$
UTH	Nondimensional flow velocity (SU) at throat
VISCC	Viscosity in the reservoir (lbm/ft-sec)

4.70 Subroutine AESØLN

DELPL(L)	Displacement thickness in cm, on Lth profile
DELSTP(L)	Displacement thickness at the throat in cm, on Lth profile
DENØM	One-fourth of the effective flow area at the throat for a channel (cm ²)
DYDX	Slope dy/dx for the first profile in a channel
DZDX	Slope dz/dx for the second profile in a channel
FAC1	Effective ordinate of first profile in a channel (cm)
FAC2	Effective ordinate of second profile in a channel (cm)

SQRTAE Square root of the effective area ratio, $\sqrt{A_e}$
X Axial coordinate in nozzle (cm)
Y Geometric ordinate of first profile in a channel (cm)
Z Geometric ordinate of second profile in a channel (cm)

4.71 Subroutine AGSØLN

A Trial value of geometric area ratio
AE Effective area ratio
AEC Constant term in equation relating the geometric and effective area ratios for a channel
AG Final solution for the geometric area ratio
DA Dummy argument in call to GEØMAR; represents X-derivative of geometric area ratio (not used)
DEL(L) Displacement thickness divided by RØ, for boundary layer on Lth profile
DELP(L) Displacement thickness (cm) for boundary layer on Lth profile
DSTP(L) Displacement thickness at the throat (cm) for boundary layer on Lth profile.
F Function which is 0 when the correct geometric area ratio has been found
FØ F value in previous step of iteration
ICØUNT Iteration counter
L DØ index over channel profiles

RNAME	Routine name for call to DUMP routine
SQRTA	Square root of geometric area ratio
UPDOWN	Indicator; 1.0 if downstream solution is desired by calling routine, -1.0 if upstream solution is desired
X	Axial coordinate in nozzle (cm)
XL	Temporary storage for previous value of X
XØ	Value of X in previous iteration
Y	Geometric ordinate of first profile in a channel (cm)
YOZO	$Y_0 Z_0$, product of profile ordinates at the throat; one-fourth of the geometric throat area in a channel (cm ²)
Z	Geometric ordinate of second profile in a channel (cm)

4.72 Subroutine AXFIT

AG	Geometric area ratio
UPDOWN	Indicator; -1.0 upstream of throat, 1.0 downstream

4.73 Subroutine AXSECT

I(J)	List of species for which cross section data are wanted (common array)
II	Index of first species in species pair
IV	Location of species II in the I array
JJ	Index of second species in species pair
JV	Location of species JJ in the I array

L	Running index for $D\emptyset$ loops
LL	Step in cross section computations where data are to be added
LQ	Position of species pair in IQ,JQ list
LQ1	Position in IQ,JQ list where additional species pairs are to be added
N	Number of species pairs added to IQ,JQ list
NN	Number of species from I array to be considered in adding data

4.74 Subroutine BLAYER

AE	Effective area ratio
AG	Geometric area ratio
AGJ	Factor proportional to r^{2j} in numerical integration from x_0 to the first flow point
AG1	Geometric area ratio in numerical integration from x_0 to the first flow point
AJ	Factor proportional to r^{2j}
ALPR	Natural logarithm of the Prandtl number at the reference temperature
AM	Mach number
AM1	Mach number in numerical integration from x_0 to the first flow point
AS	Sound speed (cm/sec)
AVCON	W/RO parameter used in averaging the correlation parameter

AX	Intermediate quality, eq. I(194d) in curvefit to shear parameter
B	Coefficient of n in linear approximation to $N(n, S_w)$, eq. I(164)
BLINTI	Value of integral I , eq. I(172), at the beginning of a step in the boundary layer solution
BM1	B-1
BX	Intermediate quantity, eq. I(194e), in curvefit to shear parameter
CC	MA_g in low Mach number region of integration from x_0 to first flow point
CØSB(L)	Cosine of angle between tangent to L th profile and the nozzle axis
C1	Intermediate quantity in quadratic solution for dM/dx
C2	Intermediate quantity in quadratic solution for dM/dx
DADX	dA_g/dx , derivative of geometric area ratio
DLM	$d \ln M/dx$, logarithmic derivative of Mach number
DMDX	dM/dx , derivative of Mach number
DUDX	Derivative of nondimensional velocity
DX	Axial distance from the previous flow point to the current one
DXI	Change in ξ in the current step

DXØ	Size of last previous "final" step, apart from steps at model points
DXP	Distance from the point before the previous flow point to the previous flow point
DXSQ	$(DX)^2$
DX2	$DX + DXP$
DYZDX(L)	Derivative of the radius of the Lth profile with respect to the axial coordinate
EK1	Coefficient A in eq. I(164)
ENPRIM	n' , eq. I(170)
FINAL	Logical flag; "false" when BLAYER is called during an intermediate calculation of the non-equilibrium integration; "true" for the final calculation of conditions at each flow point
F0	Intermediate quantity in curvefit to shear parameter, eq. I(194)
F2	Intermediate quantity in curvefit to shear parameter
F3	Intermediate quantity in curvefit to shear parameter
F4	Intermediate quantity in curvefit to shear parameter
GAMMAE	Effective specific heat ratio, $\gamma_e = a^2W/RT$, used in calculation of effective hypersonic parameter
GJ(I)	Species concentrations (moles/g)
H	Nondimensional enthalpy, hW_0/R_0T_0

HCØN	$R_0 T_0 / W_0$, factor for converting nondimensional enthalpy into cal/g
HE	Free-stream specific enthalpy (cal/g)
HF	Form factor, \hat{c}^*/θ
HFPL	HF + 1
HREF	Reference enthalpy
HTR	Incompressible form factor, H_{inc}
HTRPL	HTR + 1
HW	Enthalpy of the gas at the wall temperature
HO	Specific stagnation enthalpy (cal/g)
I	DØ index
ITYPE	Index for type of nozzle geometry ITYPE = 1 Two-dimensional nozzle ITYPE = 2 Axisymmetric nozzle ITYPE = 3 Rectangular channel
J	DØ index
K	DØ index over the cold species
L	DØ index running over the nozzle or channel profiles
LPR	Index, equal to LPRIME(L)
LPRIME(L)	Equal to 2 for L = 1, equal to 1 for L = 2
ØLDMF	Logical flag; "true" if no mole fraction has changed by 0.01 since the last flow point at which transport properties were calculated

ØMEGA	ω , viscosity-temperature exponent in eq. I (191)
ØM1	Max (1- ω , 0)
ØNE	1.
ØRDIN(L)	$\bar{\Phi}$ for the boundary layer on the Lth profile, eq. I(171)
ØRD1	A value of $\bar{\Phi}$ used in the numerical integration from x_0 to the first flow point
ØRD2	See ØRD1
P	Nondimensional pressure, p/p_0
PRPWR	(PRREF) ^{0.56}
PRW	Prandtl number at the wall temperature
R	Ratio of local nozzle radius to throat radius for axisymmetric nozzles
RA	Intermediate quantity in curvefit to Reynolds analogy factor
RAF	Reynolds analogy factor R_A in eq. I(195)
RB	Intermediate quantity in curvefit to Reynolds analogy factor
RER	Reynolds number based on R_0
RH	Nondimensional density
RMR	$\rho\mu$ ratio, $\rho_w \mu_w / \rho_e \mu_e$
RMMW	ρ_w
R1	Radius of a profile in the numerical integration from x_0 to the first flow point

R2	See R1
SIGH	Hypersonic parameter σ
SQT	Intermediate quantity in transverse curvature correction, eq. I(187)
SWFAC	$(S_w + 1) [7 + 3.4 (S_w + 1)]$
SWFAC2	$1 + 4.5 (S_w + 1)^{0.9}$
SWP1	$S_w + 1$
T	Nondimensional temperature, T/T_0
TE	Free stream temperature, T
TEP	Free stream temperature at which transport property calculations were last done
THETA1	Momentum thickness based on eq. I(185), divided by R_0
TR	T^*/T_w , ratio of reference temperature to wall temperature
TRCALC	Logical flag, .TRUE. if transport properties were calculated at the current flow point
TREF	Reference temperature ($^{\circ}K$)
TREFP	Reference temperature at which transport properties were last calculated
TRP(I)	Array equivalenced to common block /TRPROP/, to permit processing these data in a $D\emptyset$ loop
TRPSV(I)	Array used to save the transport properties at the free stream temperature
U	Nondimensional velocity, u/u_s (see US in common block /SS/)

UPRIME	Free stream velocity, u (cm/sec)
VISCR	Viscosity at the reference temperature (poise)
VISCW	Viscosity at the wall temperature (poise)
VR	VISCR/VISCW
WD	Weight factor w in calculation of $d\delta^*/dx$; eq. I(419)
WW	$e^{-\Delta x/a}$ in eq. I(217); weighting factor for \bar{n}_p in the calculation of the averaged correlation parameter, \bar{n}
X	Axial coordinate (cm)
XII	Previous value of XI
XL	Shear parameter ℓ , eqs. I(194)
XP	Previous value of x (cm)
XPXP	$X + XP$
XSAVE(I)	Temporary storage for species mole fractions SAVEC(I)
XSNV	$\min(\bar{n}, 0)$; used in place of \bar{n} in calculation of HTR to ensure the use of a nonpositive value
XTEST(J)	Mole fraction of the Jth species used in the last previous calculation of transport properties
XX	Axial coordinate variable in numerical integration from x_0
YZ(L)	Radius of the Lth profile (cm)
ZERO	0.

4.75 Subroutine BICALL

FINAL Logical subroutine argument; **.FALSE.** when **BLAYER** is called during an intermediate calculation of the nonequilibrium integration, **.TRUE.** for the final calculation of conditions at the end of each successful integration step

GJ(J) Species concentrations in moles/g

J \emptyset index over the species

4.76 Subroutine BXSECT

I First index of species pair in **V** array

J Second index of species pair in **V** array

K Value of **KQ** for step **L** of cross section computations

L Index of step in cross section computations which is currently being considered

M Number of sets of constant factors to be used in computations, for option **KQ = 14** only

MADD Index used in locating species pairs in **V** array

MM **MM + 1** and **MM + 2** are the locations of the current species pair in the **V** array

MV Index of last parameter in **V** array for step **L-1** of the computations

4.77 Subroutine COMM

ALGJ(J) $\ln \gamma_j$, natural logarithm of the **J**th species concentration

CLKF	$\ln k_f$, natural logarithm of the forward rate constant
CLKR	$\ln k_r$, natural logarithm of the reverse rate constant
CLTBF	$\ln (\sum \rho \gamma_j)$, where the sum is over the third-body species in a reaction
CLZ2	$\ln (Z2)$
CLZ3	$\ln (Z3)$
EF	ϵ_{fi} in eq. I(32lc)
ER	ϵ_{ri} in eq. I(32lc)
GJ(J)	Concentration (γ_j) of the Jth species (mole/g)
I	DØ index over the reactions
ICHG	Equal to 1 for a neutral species, equal to 2 for a charged (ionic) species
ICØUNT	Counter for number of times CØMM has been called in a run
ICYCLE	Interval in ICØUNT at which ISW5B is set equal to 1 to provide diagnostic dumps in subroutine CØMM, EXACT, RNKT, and PRTA
IPAI	Index of atomic species produced in current reaction for KTF(I) = 4
J	DØ index over the species
KF	Indicator for temperature dependence of forward reaction rate constant; for KF = 1, k_f is a function of T; for KF = 2, it is a function of T_0 . For other values of KF, see discussion of KTF input in Sec. 2.3 of Vol. II.

KR	Indicator for temperature dependence of reverse reaction rate constant; for KR = 1, k_r is a function of T; for KR = 2, it is a function of T_e
LØGC	Logical indicator for method of calculation of PICH(I); for LØGC = .FALSE., PICH(I) is computed as $P_i \chi_i$; for LØGC = .TRUE., it is calculated as $-\exp[\ln P_i + \ln(1 - \chi_i)]$ to avoid possible underflow of the exponential function in $P_i = \exp(\ln P_i)$
ØNE	1.
QELAS	Energy transfer to the electron gas as a result of elastic collisions (cal/cm ³ sec)
QF	\dot{q}_{fi} in eq. I(32lb)
QR	\dot{q}_{ri} in eq. I(32lb)
R	Effective nozzle radius for loss of resonance radiation
RCALC	Logical indicator; if .TRUE., R has already been computed for current flow point
RKF	k_f , forward rate constant
RKR	k_r , reverse rate constant
SUMF	$\sum_{j=1}^n V_{ij} (\ln \rho + \ln \gamma_j)$
SUMR	$\sum_{j=1}^n v'_{ij} (\ln \rho + \ln \gamma_j)$
TAU	Optical thickness based on R for resonance radiation
TEP	Electron temperature, T_e (°K)

TEQ	T_e^3
T6	$\sum_{j=1}^n \nu_{ij} \ln \gamma_j$
T7	$\sum \gamma_j$ over the third-body species for a reaction j
VEJ	Collision frequency factor ν_{ei} in the calculation of energy transfer to the electron gas by elastic collisions
VIC	Electron-ion elastic collision frequency (ν_{ei}) divided by the ionic species concentration γ_j
VIC1	$(8/3)\sqrt{\pi/m_e} e^4 N_0 / (2k)^{3/2}$, where m_e is the electron mass (g), e is the electronic charge (esu), N_0 is Avogadro's constant, and k is Boltzmann's constant
VIC2	$k^3 / (\pi e^6)$
VNC	$(4/3) \sqrt{8k / (\pi m_e)} N_0$
VNCC	Electron-neutral collision frequency (ν_{ej}) divided by the neutral species concentration (γ_j)
Y	Radius of first channel profile, cm
Z	Radius of second channel profile, cm
Z1	$\ln u$, natural logarithm of flow velocity in cm/sec
Z2	$R_0 T'$, where $T' = T$ for $KR = 1$ and $T' = T_e$ for $KR = 2$ ($\text{cm}^3 \text{ atm/mole}$)
Z2F	$R_0 T_0$ ($\text{cm}^3 \text{ atm/mole}$)
Z3	$\rho R_0 T'$ (see Z2)

4.78 Subroutine CXSECT

I(J) New index of species for which old index is J
J Second index of species pair
L First index of species pair
M Temporary variable

4.79 Subroutine DERIVS

CPJE Temporary storage for specific heat of the electrons
CTSAVE Temporary storage for the nondimensional gas temperature T/T_0
DELI1(L) δ_{i1}^* for the Lth profile (eq. I(424))
DELI2(L) δ_{i2}^* for the Lth profile (eq. I(424))
DELØ1(L) δ_{o1}^* for the Lth profile (eq. I(424))
DELØ2(L) δ_{o2}^* for the Lth profile (eq. I(424))
DEN Denominator in eq. I(424)
HJE Temporary storage for nondimensional enthalpy of the electron gas (H_e/R_0T_0)
I DØ index over species
ITRY Counter for iteration to obtain self-consistent solution for the boundary layer displacement thickness (Section 7.6 of Vol. I)
K DØ index over profiles
SENTE Temporary storage for nondimensional entropy of the electron gas (S_e^0/R_0)

4.80 Subroutine DSMSOL

A Matrix initially containing the coefficients and constants for a system of linear equations to be solved:

$$a_{11}x_1 + a_{12}x_2 + \dots + a_{1i}x_i = b_1$$

$$a_{21}x_1 + a_{22}x_2 + \dots + a_{2i}x_i = b_2$$

.

.

.

$$a_{i1}x_1 + a_{i2}x_2 + \dots + a_{ii}x_i = b_i$$

The coefficients a_{jk} are stored in the locations $A(j,k)$, for $j=1$ to i and $k=1$ to i . The constant terms b_j are stored in $A(j,i+1)$, for $j=1$ to i . The solution values x_k are written into the locations $A(k,i+1)$ for $k=1$ to i , that is, the locations originally occupied by the b_j .

I Number of equations in the system to be solved

J \emptyset index running from 1 to I

K \emptyset index running from 1 to I

KS Error indicator; nonzero value indicates singular matrix of coefficients

L Counter for matrix elements during rewriting of the matrix in close-packed form

LM First index for close-packed form of matrix

LN Number of columns already filled in close-packed matrix

LN1 Column index (second index) in close-packed matrix

M1 $I + 1$

N First dimension of A in the dimension statement in the calling routine

N1 N + 1

RNAME Routine name for calling DUMP routine

4.81 Subroutine DUMP

RNAME Name of routine from which DUMP was called

4.82 Subroutine DUMPEX

No local variables.

4.83 Subroutine ELCØND

I Species index

SIGMA Electrical conductivity in mhos/cm

SSIG $(e/k)^2 / \sqrt{T}$

4.84 Subroutine ELTIME

CALLED Logical indicator, initially .FALSE., set to .TRUE. when ELTIME is first called

DET Elapsed time since last printed time (seconds)

ET Elapsed time from beginning of execution (seconds)

ETM Elapsed time from beginning of execution (minutes)

ETØ ET argument value passed from calling routine

ETP Previous value of ETØ

I Elapsed time from start of execution in milliseconds

IP Control for printing the time; zero value suppressed the print

4.85 Subroutine EPART

EF Energy gained by the electron gas in the forward direction of a reaction (cal/mole)

ER Negative of the energy gained by the electron gas in the reverse reaction (cal/mole)

E0 Reaction energy to be partitioned between radiative losses and heating of the electron gas, ϵ_0 (cal/mole)

IR Reaction index in the list of reactions for the current gas model

IT Indicator (ITR) of the rule for partitioning the reaction energy in the current reaction

QF Energy lost by radiation in the forward reaction (cal/mole)

QR Negative of energy lost by radiation in the reverse reaction (cal/mole)

R0 Universal gas constant, cal/mole-°K

TE Electron temperature (°K)

THR $3R_0/2$, where R_0 is the universal gas constant in cal/mole-°K

4.86 Subroutine EOCALC

CTSAVE Variable used to save and restore the initial value of the nondimensional temperature CT

I Dø index

IL I + ISC

J DØ index
 K DØ index
 L I + ISC
 N Iteration counter for Newton-Raphson solution of equations I(233)
 P Pressure (atm)
 RNAME Routine name for calling DUMP routine
 T Temperature (°K)
 ZA Newton-Raphson correction factor $(1 + h_n^x)$ to mole fractions, eq. I(235)

4.87 Subroutine EXACT

AAA(L,M) Array of coefficients and constants for system of linear equations
 DGJ(J) $d \gamma_j / dx$
 ECØEF $1 / (\rho u R_0 T_0)$
 GJ(J) γ_j , species concentrations, moles/g
 I DØ index
 J DØ index over species
 K I - ISC
 NE Number of equations to be solved for rates of change of dependent variables during the nonequilibrium integration
 SUMTDG $\sum_{j=1}^n \frac{T}{T_0} \frac{d \gamma_j}{dx}$ for chemical nonequilibrium models;
 $\frac{T_e}{T_0} \frac{d \gamma_e}{dx} + \sum_{j=2}^n \frac{T}{T_0} \frac{d \gamma_j}{dx}$ for electronic nonequilibrium models

SI
$$-1 + v_{i-c}^* + \frac{H_i}{R_0 \bar{T}} - \sum_{j=1}^c \bar{v}_{i-c,j} \frac{H_j}{R_0 \bar{T}}$$

l
$$(u/u_s)^2/w_0$$

4.88 Function EXP

EXP Value of function

X Subroutine argument

4.89 Subroutine FINDX

A Geometric area ratio (from calling routine)

AR Geometric area ratio based on a trial value of X

DADX Derivative of geometric area ratio with respect to X (cm⁻¹)

DZDX(I) Slope of Ith profile in channel

ERA Error in geometric area ratio AR at current value of X

ERRL Smallest negative error in AR for any previous trial value of X

ERRU Smallest positive error in AR for any previous trial value of X

IENTRY Indicator of entry point; 1 for FINDX, 2 for FINDXC

MBL Index (1 or 2) of the profile whose ordinate varies most rapidly downstream of the throat in a channel

N Iteration counter

C 14

RNAME	Routine name for call to DUMP routine
UPDØWN	Indicator; -1.0 if upstream solution is desired, 1.0 if downstream solution is desired
UPPER	Logical indicator; .TRUE. if a positive ERA has been found at some trial value of X
V	Trial value of X (cm)
VL	Trial value of X corresponding to error ERRL
VØ	Trial value of X in previous iteration step
VU	Trial value of X corresponding to error ERRU
X	Final solution for X (cm)
Z(I)	Ordinate of the Ith profile in a channel

4.90 Subroutine FRØZEO

AR	Common variable, used here for the geometric area ratio
AS	Sound speed (cm/sec)
BLVAR(I)	Array equivalenced to the contents of common block /BLOUT/
BLVARØ(I)	BLVAR values at preceding flow point
BLVAR(S(I)	BLVAR values at the current regular flow point; used in interpolating at a model point
CLNMA	Logarithm of the molecular weight in the reservoir
CTSAVE	Nondimensional temperature at the regular flow point following a model point

CTSTAR	The nondimensional temperature (CT) value at the flow point preceding the throat, in solutions including the boundary layer
CT1	Second most recent trial value of CT in the iteration to find the conditions at a model point
CT2	Most recent trial value of CT in iteration to find the conditions at a model point
CXSAVE	X coordinate of the regular flow point following a model point
DELBLØ(I)	For I = 1, 2, the nondimensional boundary layer displacement thickness for the Ith profile at the preceding flow point
DELBLP(I)	For I = 1, 2, the nondimensional boundary layer displacement thickness for the Ith profile at the point before the preceding flow point
DELT	Decrement in nondimensional temperature CT in calculation of frozen throat conditions
DELTSV	Variable used to save the input value of DELT1
DELTV	Current decrement in nondimensional temperature CT
ERRX	Position error in iteration to determine the conditions at a model point
FLAG	Indicator for upstream and downstream regions, for use by the geometry subroutines. Negative value indicates an upstream position, positive value a downstream position

F1,F2,F3	Values of nondimensional mass flux in iteration to determine the frozen throat conditions
GJ(J)	Species concentrations, mole/g
I	D \emptyset index
IPASS	Indicator for iteration to take the boundary layer displacement thickness into account in the determination of reservoir conditions. IPASS = 1 causes a return to the main program after the throat has been reached
IRSTRT	Counter for number of times solution has been restarted with a lower value of W to suppress instability due to coupling of the boundary layer and the inviscid solution
IS \emptyset LN	Indicator for type of flow solution: IS \emptyset LN = 1 Frozen solution IS \emptyset LN = 2 Equilibrium solution
IS \emptyset NIC	Indicator for region of flow solution: IS \emptyset NIC = -1 Upstream region IS \emptyset NIC = 0 Sonic point IS \emptyset NIC = 1 Downstream region
ISWTCH	Counter for number of flow points beyond the throat
ITS	Index of the next specified test section diameter (TSDIAM)
J	D \emptyset index over gas species
K	D \emptyset index
LASTPT	Indicator for last point in flow solution (at CX = CXMAX)
M	D \emptyset index over channel profiles
M \emptyset DLPT	Logical indicator; if .TRUE., a model point is being calculated
N	D \emptyset index

NMPI	Counter for number of iterations to determine the conditions at a model point
ØNE	1.0
SKIP	Logical variable used to indicate that conditions at a model point have just been determined
SØLN(I,J)	Hollerith array
SWTCH	Logical indicator; if .TRUE., FRØZEQ has begun taking the displacement thickness into account in calculations of the flow-point position
S1	$-\frac{1}{w_0} \sum_{j=1}^n x_{j0} \ln x_{j0}$
S2	$-(\ln p_0)/w_0$
TSTEPC	Logical indicator for special control of temperature decrement far downstream, to prevent distance of successive flow points from becoming too large
T1, T2, T3	Nondimensional temperature values in iteration to determine frozen throat conditions
XFRAC	Distance from preceding flow point to model point, divided by distance from preceding flow point to current point
XLAST	Position of preceding flow point (cm)
XMØDEL	Model-point position (cm)
XMØD1	Position of the next model point in the geometric sequence of model points (cm)
XØ	Position of the preceding flow point (cm)
XP	Position of the point before the preceding flow point (cm)

X1 Position of the flow point corresponding to the second most recent trial value of CT in the iteration to find the conditions at a model point (cm)

X2 Position of the flow point corresponding to the most recent trial value of CT in the iteration to find the conditions at the model point (cm)

Z $F1 - F2 + F3 - F2$

4.91 Subroutine GEOM

AN Effective area ratio corresponding to the current trial value of ρ , based on the area ratio-density relation, eq. I(383)

A2 Square of the effective area ratio

ARG $2C(A_e - 1) / \alpha$

DADR $\rho_0 dA_e / d\rho$, derivative of the effective area ratio with respect to nondimensional density

DD Correction in nondimensional density

DELRHØ Absolute value of the difference between ρ / ρ_0 and the nondimensional density at the throat, based on approximate solution near throat

DELSTP Displacement thickness at the throat (cm)

D2LA $[d^2(\ln A_e) / dx^2]$, the second derivative of the logarithm of the effective area ratio at the throat

J DØ index over the profiles in a channel

N Iteration counter

NSUJ Number of upstream sections in the curvefit to a profile

ØMZ	1. - Z
ØPHA	$1 + \alpha/2$, where α is the exponent in the density-area ratio relation I(383)
RNAME	Routine name for call to DUMP
SGN	-1 if dA_e/dx is positive, +1 if dA_e/dx is negative
Y	Consistency check on x and ρ values; if positive, x and ρ data are inconsistent
Z	$(\rho/\rho_0)^\alpha$

4.92 Subroutine GEØMAR

ARATIØ	Geometric area ratio
DAB	Difference between X and the x-coordinate of the center of a circular profile section
DERIVA	dA_g/dx , derivative of geometric area ratio (cm^{-1})
DRV(I)	Slope (derivative) of ordinate for Ith profile
DYDX	Slope of ordinate for first profile, after rescaling area
DZDX	Slope of ordinate for second profile, after rescaling area
I	Index of two profiles
IENTRY	Index of entry point: 1 - GEØMAR; 2 - GMAR; 3 - GMAR2; 4 - GMAR3
ISJ	Shape index for a profile section

J	Index for the sections in a profile
K	Index for the profile section in which X lies
NSML	Number of sections in a profile minus 1
Q	Difference in ordinate between a point on a circular profile section and the circle center (cm)
RAD(I)	Ordinate of the Ith profile (cm)
RAT	Ratio of a profile ordinate to the ordinate at the throat
SN(I)	-1.0 for I = 1, + 1.0 for I = 2
SQTRSA	Square root of area rescaling factor RSA
S1(I)	Unrescaled geometric area ratio; or ratio of ordinate to ordinate at the throat for Ith profile
S2(I)	Derivative of geometric area ratio; or ratio of derivative of ordinate to ordinate at the throat for Ith profile (cm ⁻¹)
X	Axial coordinate in nozzle (cm)
Y	Ordinate of first profile (cm)
Z	Ordinate of second profile (cm)

4.93 Subroutine INGAS

CAPQ(K)	Q_k , number of gram-atoms of the kth chemical element per mole of the cold gas
CCI(I)	Intermediate variable in the calculation of the molecular weights of species from atomic weights of elements

I	Species index
IL	Index running from c + 1 to n
J	Element index
K	Index running from 1 to c
L	Index running from c + 1 to n
SUM	Dummy variable used for accumulating summations

4.94 Subroutine INIT

I	Species index in the current gas model
II	Species index in the master list of species
J	Index of additional vibrational modes in triatomic molecules
K	Number of electronic levels for a species
L	Index of the electronic levels for a species
SPRP(M, II)	Array containing species property data in common /SPEC/
T1	Reservoir temperature T_0 ($^{\circ}$ K)
T2	T_0^2
T3	T_0^3
T4	T_0^4

4.95 Subroutine KANDMU

I	Species index
IL	I-1

- (1)
- J Species index
 - L Index of property being calculated; L = 1 for translational thermal conductivity, L = 2 for viscosity
 - R Weighting factor used in computing $\bar{a}^{(\alpha)}$ from eq. I(86)
 - U Variable used to accumulate numerator in eq. I(86)
 - V Temporary variable used in computing denominator in eq. I(86)
 - W Temporary variable used in computing numerator in eq. I(86)
 - Y Temporary variable, used for accumulating denominator in eq. I(86) and numerator in eq. I(85)
 - ZK(L) ZK(1) $\cdot \sqrt{T}$ is the translational thermal conductivity in milliwatts/cm- $^{\circ}$ K, ZK(2) $\cdot \sqrt{T}$ is the viscosity in millipoise

4.96 Subroutine KINT

- (1)
- I Species index
 - II I-1
 - J Summation index for sum over species pairs with J < I
 - L Index of property being computed (see discussion of KINT in Sec. 3.28)
 - NI Species index
 - NJ Summation index for sum over species pairs with NJ > NI
- (1)

N1 N + 1

ZKINT ZKINT \sqrt{T} is the internal thermal conductivity in milliwatt/cm-°K

4.97 Subroutine LIST

CAPQ(I) Q_k, number of gram-atoms of the kth chemical element per mole of the cold gas

CØM1(I,J) Hollerith array for output (see DATA statement)

CØM2(I,J) Hollerith array for output (see DATA statement)

CXMAXI Distance beyond throat at which case will be terminated (inches)

I DØ index

J DØ index

K DØ index

KDQ1 KDIM + 1

NWXI Number of specified distances WXI from the leading edge of a wedge model

SBJ(J) b_j, eqs. I(51)

SHJAP(J) Formation enthalpy of the Jth species, cal/mole

THEVP(J) Characteristic vibrational temperature for the Jth species, °K

WØRD(I,J) Hollerith array for output (see DATA statement)

4.98 Subroutine MATINV

A Original matrix; replaced by inverse

AA Temporary storage used in interchanging columns

B Temporary storage used in interchanging rows

BB Temporary storage used in reduction of off-diagonal elements

BIG Pivotal element in a column (largest absolute magnitude)

I DØ index

IV(K) Index array for recording interchanges of rows

J Temporary storage used in interchanging IV elements; also DØ index

JA Index of current row; reset to index of next row

JB Index of preceding row

JJ Index of current row

K First index of pivotal element in a column

K1 $K + 1$

L DØ index

M Equal to MM

MM Size of matrix to be inverted

M1 $M - 1$

NN First dimension of A in dimension statement of calling routine

RNAME Routine name for calling DUMP routine

4.99 Subroutine MØDEL

ALPHA Ratio of the current estimates of the pressure p_2 and the density ρ_2 behind the shock

CATM Conversion factor from atm to dyne/cm²

CKS1 Parameter characterizing low density effects on the stagnation-point heat flux, ξK^2 (per foot)

CM1 Mean molecular weight of the gas ahead of the shock, g/mole

CTSAVE Temporary storage for nondimensional free stream temperature CT

DET Computer time used in model calculations, seconds

DISC Discriminant in quadratic solution for the pressure behind the equilibrium normal shock

ELN(I) Exponent of Lewis number in Fay-Riddell heat transfer correlation (equilibrium boundary layer, I = 1, frozen boundary layer I = 2)

EMF Temporary storage for electron mole fraction at stagnation point, to permit output of electron concentration

EPS Density ratio $\epsilon = \rho_1 / \rho_2$ across normal shock

EPSØLD Previous estimate of ϵ

EQUALS Hollerith equality sign used in output

ERRHI Most recent positive error in the nondimensional enthalpy in iterative solutions for conditions behind the normal shock and at the inviscid stagnation point

D	ERRLO	Most recent negative error in the nondimensional enthalpy
	ERZ	Current error in the nondimensional enthalpy
	ET	Elapsed time since the beginning of the run, seconds
	ETO	Elapsed time since the beginning of the run when MDEL is entered, seconds
	GJ(I)	Species concentrations, moles/g
	HCF	Enthalpy of formation of the gas mixture at the stagnation point
	HCN	Conversion factor (RT_0/W_0) from nondimensional enthalpy to enthalpy in cal/g
	HD	Nondimensional kinetic energy per unit mass of the gas ahead of the shock, $(u/u_s)^2/2$
(HE	Free-stream static enthalpy (cal/g) ; used in flat-plate calculation
	HI	Logical variable; if .TRUE., a positive enthalpy error has already been obtained in the iteration
	HR	Recovery enthalpy (cal/g) in flat-plate calculation
	HRATIO	Ratio of the dissociation enthalpy to $(h_e - h_w)$
	HSTAG	Enthalpy at the inviscid stagnation point, cal/g
	HSTAR	Reference enthalpy (cal/g) in flat-plate calculation
	HW	Gas enthalpy at the wall temperature of the stagnation point model, cal/g

H2	Nondimensional enthalpy behind the normal shock
I	D \emptyset index
IC \emptyset UNT	Iteration counter
IS \emptyset LN	Index specifying the two types of normal-shock solution (equilibrium IS \emptyset LN = 1, frozen IS \emptyset LN = 2)
ISS5P1	Number of lines of output required for the mole fractions at the inviscid stagnation point
J	Index of chemical elements
K	D \emptyset index
L	Model-shape index
LIMSC \emptyset (I)	For I = 1, 2, stored values of LMSC for the equilibrium and frozen normal shock solutions
LIM1	Lower limit of implied D \emptyset loop for output of mole fractions
LIM2	Upper limit of implied D \emptyset loop for output of mole fractions
LMSC	Number of data items from the SC \emptyset UT array to be printed for the current normal shock solution
L \emptyset	Logical indicator; if .TRUE., a negative enthalpy error has already been obtained in the iteration
N	D \emptyset index
NEL	Number of chemical elements in a species

PC	$4 \rho_1^2 u_1^2, \text{ atm-g/cm}^3$
PRW	Prandtl number at the model wall temperature
PS	Stagnation pressure, atm
PT	$p_1 + \rho_1 u_1^2, \text{ atm}$
P1	Pressure ahead of the shock, atm
P2	Pressure behind the normal shock, atm
QFRE(L)	Stagnation point heat flux times the square root of the nose radius for an equilibrium boundary layer (hemispherical model L = 1, flat-faced model L = 2), Btu/ft ^{3/2} -sec
QFRF(L)	Stagnation point heat flux times the square root of the nose radius for a frozen boundary layer (hemispherical model L = 1, flat-faced model L = 2), Btu/ft ^{3/2} -sec
QP	$p_1 T_2 / T_1$
QS	Factor used in calculating QFRE and QFRF
QSRI	Stagnation point heat flux times the square root of the nose radius based on the SRI approximation, Btu/ft ^{3/2} -sec
QUANT(I)	Hollerith names of output quantities in SCOUT array
RHØSTR	Density at reference temperature in flat-plate calculation
R1	Density ahead of the shock, g/cm ³
R1U12	$\rho_1 u_1^2, \text{ atm}$
R2	Density behind the shock, g/cm ³
SCOUT(I)	Array used for output of model conditions

SHV	Array equivalenced to common block /STAG/ to permit setting these data in the DØ loop over ISØLN
SØD(L)	Shock standoff distance divided by model radius (hemispherical L = 1, flat-faced L = 2)
SQTE	Square root of the shock density ratio, $\sqrt{\epsilon}$
SQ2F(K)	Ratio of the stagnation-point heat flux for the current model geometry to that for an axisymmetric model (two-dimensional model K = 1, axisymmetric K = 2)
THI	T ₂ value corresponding to enthalpy error ERRHI, °K
TITLE2	Hollerith data used in output
TLØ	T ₂ value corresponding to enthalpy error ERRLO, °K
TSTAR	Reference temperature in flat-plate calculation, °K
T1	Temperature ahead of the shock, °K
T2	Temperature behind the shock, °K
U1	Velocity ahead of the shock, cm/sec
VGP(L)	Velocity gradient parameter (R_p/u_1) (du_e/dx) (hemispherical model L = 1, flat-faced L = 2)

4.100 Subroutine NEWRAP

GJ(I)	Species concentrations, mole/g
I	DØ index; also J + ISC
IL	I + ISC

J	D ϕ index
K	D ϕ index
L	I + ISC
LL	ISSNR + 1
M2	ISC + 2
N ϕ DEPS	Logical indicator; .TRUE. when the gas model contains no dependent species
RNAME	Routine name for calling DUMP routine
ZB	Correction factors $1+h_n^r$ to the mole fractions, eq. I(260)
ZC	Correction factor $1+h_{c+1}^r$ to the pressure

4.101 Subroutine NEXTMP

ITS	Index of the next specified test section diameter.
ϕ NE	1.0
XM ϕ D1	Coordinate of the next model point in the geometric sequence of model points, cm
XM ϕ D2	Coordinate of the model point at the next specified test section diameter, cm
XM ϕ DEL	Coordinate of the next model point, cm

4.102 Subroutine N ϕ NEQ

AB	Temporary storage location used in interchanging rows of β_{ij} matrix during determination of matrix rank ^{ij}
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ADCH Absolute value of a $\delta\chi_i$ value

ALOGT $\ln(10000)$

BLNEAR(M) Array equivalenced to the first 10 locations of common block /BLNE/

CA BETA(M, I)/BETA(IX, I)

CAISV(I) Storage for the correct values of CAI(I) to allow resetting CAI(I) when the artificial increase of reaction rate constants has been used to prevent premature switching from the perturbation technique to the numerical integration

CI(K) Array specifying elemental composition of the gas mixture in the reservoir; CI(K) is the number of gram-atoms of the Kth element per gram of mixture

CJ Number of gram-atoms of an element per gram of mixture at the end of a successful integration step

DCA(K) $\frac{1}{(\gamma_k')^2} \sum_{j=1}^c \delta c_j A_{jk}$ in eq. I(346c)

DCHMAX Largest $|\delta\chi_i|$ for any reaction

DCHMIN Smallest $|\delta\chi_i|$ for any reaction

DCI(K) δc_k , correction in concentration of Kth chemical element required at the end of an integration step (eq. I(344), g-atom/g)

DELBLB(L) Nondimensional boundary layer displacement thickness for the Lth profile, δ_L^*/R_0 , at the start of an integration step

DELTA	Nondimensional temperature interval in the perturbation solution, $ \Delta T /T_0$
DELTSV	Input value for DELT1
DQEM	Maximum allowable change in the energy transfer to the electron gas in one integration step
DXM	Largest allowable step size for the first integration step after the switch from the perturbation solution
DXOLD	Step size Δx in the preceding integration step (cm)
DXSAVE	Temporary storage for saving and restoring the integration step size when a reduced step is used to calculate the conditions at a model point (cm)
GJ(J)	Concentration of the Jth species, γ_j , mole/g
GSQ(J)	Square of GJ(J)
HDXSAV	One-half of DXSAVE, cm
I	D ϕ index
ICOUNT	Counter for number of times a step has been restarted with reduced step size
IERR	Error exit indicator (see subroutine listing)
IF(N)	Array for storing and printing out IFAIL values for successive restarts of a step (see glossary for common block /TNERK/)
II	I + ISC
IMAX	Index of the reaction with the largest $ \delta \chi_i $

①	IRA	Counter for number of times the artificial increase in rate constants has been used to prevent premature switching from the perturbation technique to numerical integration
	ISMCP1	$n - c + 1$
	ISTEPS	Counter for number of integration steps
	ISUBD	Counter for number of times the temperature interval ΔT has been cut in half in an attempt to find a point in the perturbation solution satisfying the second inequality of eq. I(381)
	ISUPR(I)	Array of indices of the reactions which are suppressed when the decrease of a minor species is found to control the integration step
	IS5	Input value of ISW5B
①	ITAPE	Number of records written on the binary output tape since subroutine THROAT was called
	ITS	Index of the next model point defined by a specified test section diameter
	IX	Rank of the β_{ij} matrix
	I2	Index used in calculating the rank of β_{ij}
	J	Index used for various purposes
	K	Index used for various purposes
	IA	Index used in calculating the rank of β_{ij}
	L	$D\phi$ index
①	M	Index used in calculating the rank of β_{ij}

①	N	Index used in calculating the rank of β_{ij}
	NSUPR	Number of reactions suppressed when the decrease of a minor species is found to control the integration step
	ØNE	1.0
	QEØLD	Energy transfer to the electron gas at the preceding flow point, cal/cm ³ sec
	RATIØ	Ratio of the smallest and largest $ \delta\chi_i $ values
	RATIØ2	Factor by which the forward rate constant is increased in the artificial increase of a reaction rate to prevent premature switching from the perturbation technique to the numerical integration
	RNAME	Subroutine name for call to DUMP routine
①	SCD	Factor by which the step size Δx is divided when an integration step fails and has to be restarted
	SDGJ(J)	Change in the concentration γ_j of the Jth species in an integration step
	TENTH	10000.
	TEP	Electron temperature (°K)
	TP	Gas temperature (°K)
	TSZ	Temporary storage for TPRINT
	XMØDEL	Position of next model point, cm
	XMØD1	Coordinate of the next model point in the geometric sequence of model points, cm

XMSET Logical indicator, set to .TRUE. when NEXTMP is first called to determine XMØDEL

ZZZ Temporary storage

4.103 Subroutine NRMAX

DELT Decrement of nondimensional temperature

F1,F2,F3 Mass flux values in iteration to determine equilibrium sonic flow conditions

GJ(J) Species concentrations, mole/g

I DØ index over species

J DØ index over species

T1,T2,T3 Nondimensional temperature values in iteration to determine the equilibrium sonic flow conditions

Z F1 - F2 + F3 - F2

4.104 Subroutine ØUT1

ANAM Hollerith word "ARAT"

ARAT Geometric area ratio

ASØLN(I,J) Hollerith array containing names for types of solution (see DATA statement)

ASTRSK Hollerith word filled with asterisks

CHDIMS(L) Transverse dimensions of a channel at the current flow point, inches

CP Molar heat capacity at constant pressure, divided by R_0

DARAT	Derivative of the geometric area ratio
DEL	Electron density, electrons/cm ³
DELSTP(L)	Displacement thickness on the Lth profile, cm
DGJ(I)	$d \gamma_i / dx$, derivatives of species concentrations, mole/ g-cm
DLNDX(I)	$d \ln \gamma_i / dx$ for the Ith species
DNAM	Hollerith work "DIAM"
EMF	Mole fraction of the electrons
FAREA	Effective cross-sectional area of the flow at the throat, cm ²
FLX	Mass flux, lbm/ft ² -sec
FVØ2(I,J)	Array used for output of SN and XSN
GAMMA	Ratio of specific heats, $\gamma = c_p / c_v$
GJ(I)	Species concentrations, mole/g
HØUT	Enthalpy, Btu/lb
HW	Enthalpy at the nozzle wall, Btu/lb
HO	Stagnation enthalpy, Btu/lb
I	DØ index
IBL	Index such that FVØUT(IBL) is the boundary layer displacement thickness on the nozzle wall or on the broad face of a channel
IFLAG	Indicator for entry points to subroutine IFLAG = 1 Entry ØUT1 IFLAG = 2 Entry ØUT2 (ISØLN)

II	Index 7 + 10*I
INEQP1	INEQ + 1
ISØLN	Indicator for type of solution ISØLN = 1 Frozen ISØLN = 2 Equilibrium ISØLN = 3 Nonequilibrium
ISS5P1	(ISS-1)/5+1, number of lines of output re- quired for printing the species mole frac- tions
IS6	ISW6B
IX	Counter of records written on tape 8 for a given flow solution
I1	Index of the first FVØUT(I) element to be printed on a line
I2	Index of the last FVØUT(I) element to be printed on a line
I6	Counter; number of flow points printed since last output of reaction rate data
J	DØ index
K	DØ index
L	DØ index
LIM1	Index of first species mole fraction to be printed on a line
LIM2	Index of last species mole fraction to be printed on a line
PØUT	Pressure, atm
PZ	3.1415927

RØUT	Density, lb/ft ³
SØUT	Entropy, Btu/lbm-°R
STEP(I)	Hollerith array (see DATA statement)
STFAC	1/ [$\rho u(h_0 - h_w)$] , (Btu/ft ² -sec) ⁻¹
TFLØW	Total mass flow, lb/sec
TNNAM(I)	Hollerith array of variable names (see DATA statement)
TNØUT(I)	Array used for output of quantities for electronic nonequilibrium model
TØUT	Temperature, °K
TYPSELN(I,J)	Hollerith array used to indicate whether non-equilibrium solution is based on perturbation technique (INEQ = 0) or numerical integration (INEQ = 1)
UØUT	Velocity, ft/sec
VARNAM(I)	Array of Hollerith names for output variables (see DATA statement)
VARN2(I)	Array of Hollerith names for N, XSN
WNAM	Hollerith word "WIDTH"
Y	Distance from nozzle axis to nozzle surface (or to first profile in the case of a channel)
Z	Distance from axis to second profile of a channel
	4.105 <u>Subroutine PERT</u>
I	DØ index
J	DØ index

K Dø index

Z Temporary storage used in computing sums

Z1 Temporary storage used in computing sums

4.106 Function PIØMEG

I Index of entries in cross section table

PIØMEG Cross section value returned to the calling routine

TE Electron temperature, °K

4.107 Subroutine PRØP

J Dø index over species

RNAME Routine name for call to DUMP

S4 $\frac{I}{R_0} \sum_{j=1}^n \gamma_{j0} S_j^{\circ}$

4.108 Subroutine PRTA

AS Speed of sound, cm/sec

4.109 Subroutine PUTQIN

I First index of species pair

IDBG Count to determine when debug output is produced

J Second index of species pair

K Option KQ to be used in step L of the computations

KK Type of cross section; first index of Q(KK,I,J)

- L Index of current step in cross section computations
- LQ Index of species pair in IQ,JQ array
- LQ1 Index of first species pair in IQ, JQ array to be used in step L of computations
- LQ2 Index of last species pair in IQ, JQ array to be used in step L of computations
- M Index of first parameter in V array to be used in step L of computations
- $\emptyset M(KK)$ Variable for temporary storage of cross sections computed in current step. $\emptyset M(1) = \bar{\Omega}(1,1)$, $\emptyset M(2) = \bar{\Omega}(2,2)$, $\emptyset M(3) = B * \bar{\Omega}(1,1)$
- WR \emptyset TE WR \emptyset TE = .TRUE. indicates that debug output has been produced in the current call to PUTQIN
- X Electron partial pressure in atmospheres
- 4.110 Subroutine QC \emptyset UL
- C Control variable. For $C > 0$, $\ln(f \wedge)$ is computed from eqs. I(100); for $C \leq 0$, $\ln(f \wedge)$ is set to 1
- $\emptyset M1$ $0.8Q_C$ (see eq. I(100b))
- X Electron partial pressure in atmospheres
- Y $20.91Y = (f \wedge)^2$ (see eqs. I(100c) and I(100d))
- 4.111 Subroutine QEX
- A Coefficient A in eq. I(104) for the exchange cross section

①

B Coefficient B in eq. I(104) for the exchange cross section

$\sigma_M(K)$ Contribution to cross sections from option;
 $\sigma_M(1) = \bar{\Omega}(1,1)$ $\sigma_M(2) = \bar{\Omega}(2,2)$
 $\sigma_M(3) = B * \bar{\Omega}(1,1)$

V(M) Input parameters VV for option

Y Quantity $A - \frac{1}{2} B \ln(T/M)$

ZM Molecular weight

4.112 Subroutine QEXP

A Coefficient A/k in $^{\circ}K$ for the exponential potential

ALPHA Quantity α

N1 Position of first entry in tabulated collision integrals for the exponential potential

$\sigma_M(K)$ Computed cross section values (see subroutine QEX)

RHO Characteristic length ρ for the exponential potential in \AA

V(M) Input parameters VV for option

②

4.113 Subroutine QINTRP

A Value of independent variable to be used
in interpolation

B(J) Interpolated value of Jth dependent var-
iable

I Interpolation is performed between entries
I and I + 1 of the table

J Index of dependent variable

N Number of entries in table

N1 Location of first entry in table

N2 N2 + 1 is the location of the last entry
in table

4.114 Subroutine QLJ

EPSLN Well depth parameter ϵ/k for the Lennard-
Jones potential, in $^{\circ}\text{K}$

N1 Position of first entry in tabulated
cross sections for the Lennard-Jones
potential

$\phi\text{M}(K)$ Computed cross section values (see
subroutine QEX)

SIGMA	Collision diameter σ for the Lennard-Jones potential, in \AA
TSTAR	Nondimensional temperature, kT/ϵ
V(M)	Input parameters VV for option
4.115 <u>Subroutine QMIX</u>	
I	First index of species pair
J	Second index of species pair
K	Type of cross section, i.e., first index of $Q(K,I,J)$
IQ	Index of species pair in IQ, JQ array
IQ1	Index in IQ, JQ array of first species pair to be used in computation
IQ2	Index in IQ, JQ array of last species pair to be used in computation
SQT(K,I)	$\frac{1}{2} \sqrt{Q(K,I,I)}$
4.116 <u>Subroutine QREPP</u>	
ITL	Location in \emptyset MEGAL array where data are stored
\emptyset M(K)	Computed cross section values (see subroutine QEX)
VV(M)	Input parameters VV for option
4.117 <u>Subroutine QSAME</u>	
C	Constant by which previously computed cross sections are multiplied
I	First index of species pair for previously computed cross sections

J Second index of species pair for previously
computed cross sections

$\emptyset M(K)$ Computed cross section values (see subroutine
QEX)

VV(M) Input parameters VV for option

4.118 Subroutine QTAB

A Constant multiplying factor for tabulated
cross sections

N1 Location of first entry in tabulated cross
section data

NL Number of entries in tabulated cross section
data

$\emptyset M(K)$ Computed cross section values (see subroutine
QEX)

V(M) Input parameters VV for option

4.119 Subroutine Q11

LQ1 Location in IQ, JQ list of the first species
pair to which the ramp function is to be ap-
plied

LQ2 Location in IQ, JQ list of the last species
pair to which the ramp function is to be applied

T0 Temperature in $^{\circ}K$ beyond which cross sections
are set to zero

T1 Temperature in $^{\circ}K$ beyond which cross sections
are unchanged

VI(M) Input parameters VV for option

VV(M) Input parameters for subroutine Q14

4.120 Subroutine Q12

I First index of first previously computed species pair

J Second index of first previously computed species pair

K First index of second previously computed species pair

L Second index of second previously computed species pair

M Index indicating cross section type

\emptyset M(M) Computed cross section values (see subroutine QEX)

VV(N) Input parameters VV for option

4.121 Subroutine Q13

C Constant by which cross section is multiplied

I First index of species pair

J Second index of species pair

K1 Type of cross section to be computed

K2 Type of cross section to be used in computation

IQ Location of species pair in IQ, JQ list

IQ1 Location in IQ, JQ list of first species pair to be used

⊕
LQ2 Location in IQ, JQ list of last species pair
to be used

VV(M) Input parameters VV for option

4.122 Subroutine Q14

C2 Constant applied to $\bar{\Omega}$ (2,2)

C3 Constant applied to $B^* \bar{\Omega}$ (1,1)

I First index of species pair

J Second index of species pair

LQ Location of species pair in IQ, JQ list

LQ1 Location in IQ, JQ list of the first species
pair to be used with the set of multiplying
factors

⊕
LQ2 Location in IQ, JQ list of last species pair
to be used with the set of multiplying fac-
tors

VV(M) Input parameters VV for option

4.123 Subroutine RADIUS

AG Geometric area ratio

AGJ Factor proportional to r^{2j} in eq. I(171)

ITYPE Indicator for type of nozzle geometry
 ITYPE = 1 Two-dimensional nozzle
 ITYPE = 2 Axisymmetric nozzle
 ITYPE = 3 Channel

L Index of profiles (1 for a nozzle, 1 or 2 for
a channel)

M	NPR(L)
NPR(L)	2 for L = 1; 1 for L = 2
R	Distance from axis to profile
RATIØ	R/R0
X	Axial coordinate in nozzle or channel, cm
YZ(L)	Distance from axis to Lth profile in a channel
	4.124 <u>Subroutine READ</u>
AL	Number of atoms of the Kth element per molecule of the Ith species, as obtained from the SPRP array
ASYM(IZ)	Hollerith chemical symbol for the element with atomic number IZ
CAPQ(I)	Number of gram atoms of the Ith chemical element per molecule of the cold gas
CP(I,J)	Array for addressing the channel data in common block /CHAN/
CSMW	Molecular weight of one of the cold species
DAL	Error in rounding A to an integer LP
DUM(I)	Array for zeroing common block /TNCE/
EIENAME(K)	Hollerith name for the Kth element in one of the cold species
EPRP(I,J)	Array for addressing the element data in common block /ELEM/

EXPMP	Exponent used in computing the factor (FACMP) used to generate a geometric sequence of distances x at which model calculations will be done
FNMP	Number of model points in geometric sequence, minus 1
GPRP(I,J)	Array for addressing the gas model data in common block /MIXT/
HWINØV	Estimate of stagnation enthalpy based on Wino- vich equation, Btu/lb
I	DØ index
IES	Index of an element in a species, in the mas- ter list of elements
IGMV	Number of electronic levels in a species
IGS	Input value of IGAS
II	Index used in place of subscripted integer variables
IPRFL	Index used in place of subscripted integer variable, NPRØFL(J)
ISH	A value of ISHAPE(K,J)
ISKUR	Index of a third-body species in the master list of species
IUPDI	Initial value of IUPD
IZ	Atomic number of an element defined in the input using EEPRP
J	DØ index
JDP1	JDIM + 1

JJ	Index used in place of a subscripted integer variable, JCS(I)
JPS	Index of a species on the product side of a reaction, in the master list of species
JRS	Index of a species on the reactant side of a reaction, in the master list of species
K	DØ index
KK	Index of an element in a cold species, in the master list of elements
L	DØ index
LL	Index used in obtaining the PARAM array from the data stored in the ZPRP array
LP	Integer obtained by rounding AL
NATØM	Number of atoms of an element in a molecule of a species
NELS	Number of elements in a species
NK	Number of third body species for a reaction
NØZZLE	Index of the nozzle profile in the precoded geometry fits
NPSS	Number of product species for a reaction
NRSS	Number of reactant species for a reaction
NSECTD(J)	Number of downstream sections in the profile curvefit for the Jth profile of a channel or nozzle
NSECTJ	Index used in place of dimensioned integer variable, NSECT(J)

NSM1	NSECT(J) - 1
ØNE	1.0
PLUS	Hollerith + sign
RPRP(I,J)	Array for addressing reaction data in common block /REAC/
SBJ(J)	b_j ; eqs. I(51)
SETIGS	Logical indicator; .TRUE. if automatic air model selection option is in effect
SHAPD	Array of Hollerith data for describing the shapes of nozzle profile sections
SHJAP(I)	Formation enthalpy of the Jth species, cal/mole
SPNAME(I)	Array of nominal species names for species defined in the input
SPRP(I,J)	Array for addressing species data in common block /SPEC/
SY1(K)	Array of Hollerith left parentheses
SY2(K)	Array of Hollerith right parentheses
THEVP(I)	Vibrational temperature for the Ith species, °K
TNEP(I,J)	Array for addressing thermal nonequilibrium data in common block /TNE/
WØRD(I,J)	Hollerith data for output of the words "two-dimensional" and "axisymmetric"
WØRD2(I)	Hollerith data for output of the words "gap" and "diam"

ZPRP(I,J) Array for addressing the geometry data in common block /NØZZ/

4.125 Subroutine RESET

I Elapsed time since the beginning of the run in milliseconds

ICPU CPU time remaining before TIME.GØ, milliseconds

IN Entry point indicator: IN = 1, RESET; IN = 2, TIME

IZERØ CPU time remaining before TIME.GØ at call to RESET, milliseconds

4.126 Subroutine RESTMP

AM(I,J) Matrix of coefficients for system of equations in quadratic interpolation for the reservoir temperature

BM(I) Constant terms for system of equations in quadratic interpolation for the reservoir temperature

CØNST $R_0(m*/p_0)^2$, in which R_0 is the universal gas constant, m_* the known sonic mass flux, and p_0 the reservoir pressure, all expressed in cgs units

CTAPP Previous estimate of reservoir temperature, °K

CTSAVE(I) For I = 1 to 3, successive estimates of the reservoir temperature, the value for I = 1 being the most recent, °K

CTSV Temporary storage for the reservoir temperature estimate used in the current iteration, °K

DCT The difference between the two most recent estimates of the reservoir temperature, °K

DCTAP	Absolute value of the difference between the current and improved estimates of the reservoir temperature, °K
DCTAPØ	DCTAP in the previous iteration step, °K
DMC	Intermediate quantity in quadratic interpolation for the reservoir temperature, °K
DL,D2	Intermediate quantities in linear interpolation for the reservoir temperature, °K
F(I)	For I = 1 to 3, successive estimates of the quantity $W_0 \cdot (SM)^2$, the value for I = 1 being the most recent; W_0 is the molecular weight in the reservoir and SM the nondimensional sonic mass flux
H	Calculated value of reservoir enthalpy, cal/g
HERR	Allowable error in calculated stagnation enthalpy cal/g
HØ	Calculated stagnation enthalpy in previous iteration step, cal/g
I	DØ index
J	Index for shifting data in CTSAVE and F arrays
KS	Indicator for singularity of AM matrix
N	Iteration counter for main iteration
N2	Iteration counter for temperature calculation in enthalpy option
N3	Iteration counter for pressure calculation in enthalpy option

PCGS	Reservoir pressure, dyne/cm ²
PERR	Allowable relative error in reservoir pressure
PIAST	Previous estimate of reservoir pressure, atm
RGAS	Universal gas constant in cgs units
RNAME	Hollerith name for calling DUMP routine
SMERR	Allowable error in sonic mass flux, g/cm ² sec
SMP	Calculated sonic mass flux, g/cm ² sec
TERR	Allowable relative error in reservoir temperature
TLAST	Previous estimate of reservoir temperature, °K

4.127 Subroutine RNKT

DGJ(J)	$d \gamma_j / dx$, derivatives of species concentrations, mole/g-cm
DQMRK	Twice DQMAX; criterion value for allowable changes in energy transfer to the electrons, QDPE, cal/cm ³ sec
FK(J,K)	Array equivalenced to F1(J), F2(J), F3(J), F4(J)
F1(J)	f_1 , derivative of the Jth dependent variable at the start of the integration step
F2(J)	f_2 , derivative of the Jth dependent variable at x_2, y_2 (eq. I(394a))
F3(J)	f_3 , derivative of the Jth dependent variable at x_3, y_3 (eq. I(394b))

F4(J) f_4 , derivative of the Jth dependent variable at x_4, y_4 where y_4 is given by I(394c) or I(405)

GJK(J,K) Array equivalenced to GJ1(J), GJ2(J), GJ3(J), GJ4(J)

GJ1(J) y_1 , the Jth dependent variable at the start of the integration step

GJ2(J) y_2 , the Jth dependent variable at x_2 (eq. I(394a))

GJ3(J) y_3 , the Jth dependent variable at x_3 (eq. I(394b))

GJ4(J) y_4 , the Jth dependent variable at x_4 (eq. I(394c) or I(405))

IFØR Indicator for type of formula used to compute GJ4(J); IFØR=1 for I(394c), IFØR=2 for eq. I(405)

ISTMNT Indicator for locations in subroutine at which diagnostic dumps are written (see listing)

J Index of dependent variables:
 $J = 1$ to n γ_j
 $J = n + 1$ T_j
 $J = n + 2$ T_e
 $J = n + 3$ h_0

K DØ index

LIM Number of dependent integration variables

P(J) Parameter ($-P \Delta x$) for the Jth dependent variable, eq. I(402)

QDPEB Electron energy transfer (QDPE) at the start of the integration step, $\text{cal/cm}^3 \text{ sec}$

SDGJ(J) Change in γ_j over an integration step, mole/g

SDQ(J)	Change in Jth dependent variable over the integration step
S1	Temporary storage in computation of I(398a) for small $P \Delta x$
S2	Temporary storage in computation of I(398b) for small $P \Delta x$
S3	Temporary storage in computation of I(398c) for small $P \Delta x$
TE	Logical indicator, .TRUE. for chemical non-equilibrium models, .FALSE. for electronic nonequilibrium models
T1	Temporary storage in computation of I(398a) for small $P \Delta x$
T2	Temporary storage in computation of I(398b) for small $P \Delta x$
T3	Temporary storage in computation of I(398c) for small $P \Delta x$
XX	$\log_{10}(f_3 - f_2)$
X1	F_1 (eq. I(398a))
X2	F_2 (eq. I(398b))
X3	F_3 (eq. I(398c))
X4	$f_2 - f_1$
Y(K)	Array equivalenced to Y1, Y2, Y3, Y4
YY	$\log_{10}(f_2 - f_1)$
Y1	$f_1 + P_{Y1}$ in I(401)

Y2 $f_2 + Py_2$ in I(401)

Y3 $f_3 + Py_3$ in I(401)

Y4 $f_4 + Py_4$ in I(401)

4.128 Subroutine SHOCK

ALP The quantity $\sin^2 \sigma - B$, where σ is the shock angle

B The quantity $-\left[\frac{M^2+2}{M^2} + \gamma \sin^2 \alpha\right]$, where M is the free-stream Mach number, γ the specific heat ratio, and α the body angle

BETA The quantity $ALP \cdot \sin^2 \sigma + C$

BMF Free-stream Mach number, M

BM2 M^2 , square of the Mach number

BM4 M^4

C The quantity $\frac{2M^2+1}{M^4} + \left[\frac{1}{4}(\gamma+1)^2 + \frac{\gamma-1}{M^2}\right] \sin^2 \alpha$

CD Cosine of the body angle, $\cos \alpha$

C1 $\gamma + 1$

C2 $\gamma - 1$

C3 2B

D $-\cos^2 \alpha / M^4$

DEL The body angle α ; inclination of the wedge surface to the direction of free-stream motion (degrees)

DELR α in radians

DIS $\sqrt{(ALP)^2 - 4 \cdot BETA}$

FL(I) Array used for communicating results to the calling routine

FL(1) shock angle

FL(2) Mach number behind shock

FL(3) Density ratio, ρ_2/ρ_1

FL(4) Static pressure ratio, p_2/p_1

FL(5) Static temperature ratio, T_2/T_1

FL(6) Total pressure ratio, P_{t2}/P_{t1}

FL(7) Ratio, T_{stag}/T_1

FPU Derivative dF/du of the function $F(u)$

FU $F(u) = u^3 + Bu^2 + Cu + D$

GAM Specific heat ratio γ

IERR Argument whose value is set to 2 when Newton-Raphson iteration to find the shock angle does not converge

KIT Iteration counter

RT $\sin^2 \sigma$, where σ is the shock angle

SD $\sin \alpha$, where α is the body angle

UG Previous trial value of u in solution of cubic equation $F(u) = 0$

UN Current trial value of u in solution of cubic equation $F(u) = 0$

U1 First solution for $\sin^2 \sigma$

U2 Second solution for $\sin^2 \sigma$

V1 $M^2 \sin^2 \sigma$

4.129 Subroutine SIMQ

A(I) Matrix of coefficients for the system of equations to be solved, stored columnwise with no gaps

B(I) Constant terms for the system of equations to be solved; these data are replaced by the solution values

BIGA Pivotal (largest) element in a column

I Row index

IA Index of matrix A in back solution

IB Index of recomputed array B in back solution, containing solution values

IC Index of array B in back solution

IJ Index in search for pivotal element in column

IMAX Index of pivotal element in column

IQS Index in elimination of variables

IT Index used for several purposes

IX $D\emptyset$ index in elimination of variables

IXJ Index in elimination of variables

IXJX Index in elimination of variables

I1,I2 Indices in interchange of rows

J Column index

JJ Index of the first element in a column of the reduced matrix

JJX	Index in elimination of variables
JX	DØ index in elimination of variables
JY	J + 1
K	DØ index in interchange of rows
KS	Error indicator; set to 1 if matrix of coefficients is singular
N	Number of equations to be solved
NY	N - 1
SAVE	Temporary storage location used in interchanging rows and eliminating variables
TØL	Criterion for value of pivotal element in test for singular matrix

4.130 Subroutine STUNTS

ASTRSK	Hollerith word filled with asterisks
DELTT(I)	Array of preset temperature intervals for table of species thermal properties, °K
H00	Heat of formation of a species in kcal/mole
I	DØ index; also value of IQ(IQ); also value of ISEQ(L)
ICARD	Counter of punched cards containing cross section data
II	Index of preset temperature intervals for thermal property table
IIQJ	Index of the first species of a pair of species whose cross section is computed in the Ith step of the transport property calculation

IT	Counter for lines in the thermal property table with the same value of the temperature increment
IZ	Number of lines in thermal property table
J	D \emptyset index; also JQ(LQ)
JJQJ	Index of the second species of a pair of species whose cross section is computed in the Lth step of the transport property calculation
K	D \emptyset index
KK	KQ(L)
L	Index of steps in the cross section calculation
LIM	1 for species with no thermo fit; 2 for species with a thermo fit
LIM1	1 for species with physical model data; 2 for species without such data
LQ	D \emptyset index
LQ1	Index of the first species pair (in the IQ,JQ arrays) to which the Lth step of the edited cross section computation is applied
LQ2	Index of the last species pair (in the IQ,JQ arrays) to which the Lth step of the edited cross section computation is applied
M	D \emptyset index
MV1	Index of the first parameter (in the V array) used in the Lth step of the edited cross section computation
MV2	Index of the last parameter (in the V array) used in the Lth step of the edited cross section computation

NDT(I)	Number of lines in the thermal property table for which the Ith preset temperature increment is used
NNQI	Number of species pairs to which the cross section calculated in the Ith step of the calculation is applied
TPRØP(I,K,M)	Computed thermal properties for the Ith line of the table; M = 1 contains the results of the physical model, M = 2 the thermo-fit results (if any) K = 1 Chemical potential K = 2 Enthalpy K = 3 Specific heat K = 4 Entropy
TT(I)	Temperature value for the Ith line of the thermal property table, °K
TV	Temperature value, °K
	4.131 <u>Subroutine THERM</u>
CCPTF	Nondimensional specific heat based on the thermo fit
CLGT	Natural logarithm of the nondimensional temperature, $\ln(T/T_0)$
E1	Logical variable, .TRUE. if the routine was entered through the entry point THERM1, otherwise .FALSE.
J	Species index
K	Index of vibrational states in a linear triatomic species
M	Index of the electronic states in a species

MIX	Logical variable, set to .TRUE. if the species thermal properties are to be calculated by "mixing" results from the physical model and the thermo fit
N	Number of electronic states in a species
SHTF	Nondimensional enthalpy based on the thermo fit
STF	Nondimensional entropy based on the thermo fit
SZ1	Temporary storage for a value required in the entropy calculation for a linear triatomic species
S1	See eq. I(56a)
S2	See eq. I(56c)
S3	See eq. I(56d)
WTHØ	Weight of the physical model results in the "mixed" calculation of species properties
WTF	Weight of the thermo fit results in the "mixed" calculation of species properties
WW	$(S2)^2$
XMTF	Chemical potential based on the thermo fit results
XX	Dummy variable used in calculations of electronic state contributions to species properties
ZT	Difference between the temperature and the switchover temperature CTMXXI from the physical model to the thermo fit, °K
Z1-Z9	Storage for intermediate values in calculations of the vibrational contributions to species properties

4.132 Subroutine THRØAT

AG Geometric area ratio
I DØ index
ØNE 1.0
XX Dummy argument for call to AGSØLN

4.133 Subroutine TRANSP

C(I,J) Molecular weight dependent factor used in transport calculations (see eqs. 171)
CMSAVE Mean molecular weight W for the mixture
CPTØT Nondimensional frozen specific heat at constant pressure for the mixture, $W c_{pf}/R_0$
CSAVE(I) Array for saving previously computed values of the nondimensional specific heats CCPJ(I)
CTSAVE Variable for saving previous value of the temperature ratio CT
I Species index
J Species index
J1 J - 1
P Gas pressure in atmospheres
S \sqrt{TTAB}
SSIG Factor used in computation of electrical conductivity, $(e/k)^2/\sqrt{T}$
TTAB Temperature in °K for which transport calculations are to be made

ZK(L)	See eq. (173)
ZKINT	See eq. (173)
ZM1(I)	Factor proportional to the molecular weight of species I (see eq. (171))
4.134 <u>Subroutine WEDGE</u>	
A	$(\gamma + 1)/2$, where γ = specific heat ratio
AE	Free-molecule accommodation coefficient for heat transfer
ALMIN	Minimum angle of attack for calculation of free-molecule heat flux (0.01 radian)
ALPHA	Angle of attack in radians
AM	Free-stream Mach number M
AMD	Double-precision Mach number for call to SHOCK routine
AM2	M^2
AND	Input angle of attack in degrees or ANMIN, whichever is larger
ANMIN	Minimum angle of attack for oblique shock calculation (10^{-4} degree)
APR(IA)	Static pressure ratio, p_2/p_1 , based on the oblique shock calculation for the IAth angle
AST(J)	Hollerith array for marking heat fluxes which are at the free-molecule limit with an asterisk
ASTQ(IT)	Hollerith blank or asterisk for ITth column of output table
A2	A^2

A4	A^4
A5	A^5
A7	A^7
C	$[\mu(T^*)/\mu(T_1)] \cdot [T_1/T^*]$; see Sec. 8.2.2, Vol. I
CAPGAM	Angle-of-attack parameter Γ (eq. I(48lc) or I(50L))
CH	Heat transfer coefficients, C_H
CHD	Coefficient $0.332 A \epsilon \tau N^2 t$ for calculating C_H from S^*
CHDFAC	Coefficient $0.332 A \epsilon$
CPGM	Single-precision value equal to Γ
DELSTW(IT)	Boundary layer displacement thickness at the ITth point along the wedge
DFAC	Coefficient $\gamma^{-k^3/8} A^5 \epsilon$
DFAK	Quantity $DFAC \cdot t/(\tau N)^4$
EN	Parameter $N = M \sqrt{C/Re_t}$
EPS	Quantity $\epsilon = (\gamma - 1)/(\gamma + 1)$
EPS2	ϵ^2
ERFV	Quantity $1 + \operatorname{erf}(S \sin \alpha)$ used in calculating free-molecule limit to heat flux
EXPV	Quantity $e^{-(S \sin \alpha)^2}$ in free-molecule heat flux calculations

FINAL	Logical variable, set to .TRUE. when calculations have been done for all of the specified distances from the leading edge for given angle of attack and nose radius
FL(K)	Array used by subroutine SHØCK to communicate its results to WEDGE. See glossary of symbols in SHØCK for definitions of the elements of FL
GAMMA	Specific-heat ratio, γ
GAMD	Double-precision value of GAMMA, for call to SHØCK routine
GFAC	Factor used in calculating Γ ; equal to $k/2A^2\epsilon M$ for the modified Cheng-Kemp theory, and to $\gamma k/2 A^2\epsilon$ for the unmodified theory
GFAK	$GFAC/(\tau N)^2$
GFC	Coefficient $k/2A^2\epsilon$ in GFAC
GMW	Mean molecular weight of gas, g/mole
G2	γ^2
G3	γ^3
HRATIØ	Ratio h_w/h_0 of enthalpy at the wall to the free-stream stagnation enthalpy
HW	Gas enthalpy at the wall temperature, cal/g
H0	Free-stream stagnation enthalpy, cal/g
IA	Index of angles of attack
IAMAX	Index of angle of attack at which classical oblique shock solution gives a detached shock

IERR	Error indicator for classical oblique shock solution, set in SHOCK routine: 0 Detached shock 1 Attached shock - normal case 2 Convergence failure
IFM	Control variable; reset from 0 to 1 if any heat flux value is reduced to the free-molecule limit
IR	Index of leading edge radii
IS9	Absolute value of ISW9B
IT	Index of columns in the output table
ITAB	Index of blocks in the output table
ITH	Index equal to 1 for the modified Cheng-Kemp theory, equal to 2 for the unmodified theory
ITHS	Value of ITH before resetting because of IERR = 0
J	Implied DØ index in WRITE statements
NW1	Index of points in the uniform sequence of distances from the leading edge
NW2	Index of specified distances from the leading edge
ØMEGA	Parameter Ω , eq. I(502)
PFAC	Quantity $4 A^4 \epsilon^2 M^2 / \gamma k^2$
PFAK	Coefficient PFAC $\cdot (\tau N)^4$ in formula for pressure on wedge
PI	Value 3.1415927
PINF	Free-stream pressure, atm

PRATIØ	Ratio of pressure on wedge to free-stream pressure, p_w/p_1
PW:(IT)	Pressure on the wedge at the ITth distance from the leading edge, atm
QFM(IA)	Free-molecule limit to the heat flux for the IAth angle of attack, Btu/ft ² -sec
QFMC1	Coefficient $0.08885 p_{1a_e} \sqrt{RT_1/2} \pi W$ in calculation of free-molecule heat flux, eq. I(512)
QFMC2	Coefficient $S^2 + [\gamma - \frac{1}{2} (\gamma+1) T_w/T_1]/(\gamma-1)$ in free-molecule heat flux calculation, eq. I(512)
QWW(IT)	Heat flux to the wedge at the ITth distance from the leading edge, Btu/ft ² -sec
REPF	Reynolds number per foot
RGC	Gas constant (8.314×10^7 erg/mole °K)
RHØINF	Free-stream density, ρ_1
RNAME	Hollerith routine name for call to subroutine DUMP
RU2	Dynamic pressure $\rho_1 u_1^2$
S	Free-molecule flow parameter $M\sqrt{\gamma/2}$
SQT	Quantity $\sqrt{zz' + \Omega \zeta}$ in modified Cheng-Kemp calculation; $\sqrt{zz'}$ in unmodified calculation
SQTPI	$\sqrt{\pi}$
SSA	$S \sin \alpha$
SSASQ	$(S \sin \alpha)^2$

TAU	Parameter $\tau = 0.664 + 1.73 h_w/h_0$
TAUN	τN
TAUN2	$(\tau N)^2$
TAUN3	$(\tau N)^3$
TAUN4	$(\tau N)^4$
TAUN6	$(\tau N)^6$
TINF	Free-stream temperature, T_1
TN	Diameter of leading edge, inch
TRATIØ	Ratio of wedge surface temperature to free-stream temperature
TREF	Reference temperature, $(1 + 3 h_w/h_0) T_0/6$, where T_0 is the free-stream stagnation temperature behind a frozen normal shock
UNIF	Free-stream velocity, u_1
UN(J)	Hollerith array for labelling the output as being based on the modified or unmodified Cheng-Kemp theory
VMUINF	Free-stream viscosity, lb/ft-sec
WK2	Square of leading-edge drag coefficient, k^2
WK3	k^3
WK4	k^4
WKV1	Next distance from the leading edge in the specified uniform sequence, inch

WXV2 Next separately specified distance from the leading edge, inch
XC Coordinate x parallel to the free-stream flow for a point on the surface of the wedge at a distance XW(IT) from the leading edge; $XC = XW(IT) \cdot \cos \alpha$, inch
XM Distance from the leading edge up to which merging effects are important (inch); see eq. I(510a)
XS Distance from the leading edge beyond which the strong-interaction approximation breaks down (inch); see eq. I(510b)
XW(IT) The ITth specified distance from the leading edge, inch
YS(IT) Shock ordinate at the ITth distance from the leading edge, inch
YSFAC Coefficient $\gamma^2 k^3 / 8 A^4 \epsilon$ in calculation of shock ordinate
YSFAK Coefficient $YSFAC \cdot t / (\tau N)^4$ in calculation of shock ordinate
Z Nondimensional shock ordinate z; eq. I(481a)
ZERØ 0.0
ZETA Nondimensional coordinate ζ parallel to free-stream flow; see eq. I(481b)
ZFAC Coefficient $16 A^7 \epsilon / \gamma^3 k^4$
ZFAK Coefficient $ZFAC \cdot (\tau N)^6 / t$
ZTA(IT) ζ at the ITth distance from the leading edge
ZZP zz'
ZZPP $(zz')'$

ZZPS	z..' as a single-precision value
	4.135 <u>Subroutine WESØLN</u>
B0	Coefficient $\sqrt[3]{6}$ in initial approximation to λ
B1	Coefficient $\sqrt{3}$ in initial approximation to λ
CALLED	Logical variable used to bypass resetting of the "C" coefficients after the first call to WESØLN
CAPGAM	Angle of attack parameter, Γ
CGSQ	Γ^2
C1	1. The "C" symbols are coefficients in the Cheng-Kemp analytical solution of the Cheng equation for $\Gamma = 0$
C2	2.
C3	1/3
C4	4.
C9	9.
C32	1.5
C103	10/3
C229	22/9
C463	46/3
C769	76/9
DLDZT	$d\lambda/d\zeta$ for Newton-Raphson iteration

N	Iteration counter
P1	$\zeta^{1/3}$
P2	$\zeta^{1/2}$
P12	$\zeta^{2/3}$
Q1	$1 + \sqrt{\lambda}$
Q2	$(1 + \sqrt{\lambda})^2$
Q3	$(1 + \sqrt{\lambda})^3$
Q4	$(1 + \sqrt{\lambda})^4$
Q5	$\ln(1 + \sqrt{\lambda})$
SQTL	$\sqrt{\lambda}$
XL	λ
Z	Nondimensional shock ordinate, z
ZETA	Nondimensional coordinate ζ parallel to the direction of free-stream flow
ZT	ζ calculated using trial value of λ
ZZP	zz'
ZZPP	$(zz')^2$

4.136 Subroutine XSECT

Because a number of symbols are used for different purposes in different sections of subroutine XSECT, the symbols used in each section of the subroutine are defined separately below.

Statements to Statement 40

- I(J) Revised index for species J in master species list (common variable)
- J Index of species in master species list
- JJ Revised index for species not included in transport calculations
- K Count of species
- L Revised index for species included in transport calculations

Statements 40 to 110

- J Species index
- K Index of steps in the cross section calculation in the order in which they are carried out
- KIJ Index in packed lists of species pairs to which edited cross section data are applied
- KM Index of parameters for a single step of the cross section calculation
- KV Index in packed list of parameter values for all steps of the cross section calculation
- L Species index; step of cross section computations
- LL First index of species pair
- LQ NQ(L)
- M Index of the Kth step in the cross section calculation, equal to ISEQ(K)

MMV	Total number of parameters in V array
NK	Index of the option to be used in the Kth step; also, number of parameters used in step
	<u>Statements 110 through 190</u>
I(J)	Revised index for species J in master list (common variable)
II	First index of unused species pair
III	II-1
J	Second index of species pair referenced in V array; species index in master species list
JJ	Index of species added to cross section computation
K	Index in master species list of species added to cross section computations
L	Index of step in cross section computations
LL	First index of species pair referenced in V array
IQ	Location of species pair in IQ, JQ array
IQ1	Location in IQ, JQ array of first species pair for step
IQ2	Location in IQ, JQ array of last species pair for step
MV	Dummy variable (not used)

Statement 190 to Card XSE 213

IV First index of species pair computed with
option KQ = 10

J Second index of species pair to be moved

JV Second index of species pair computed with
option KQ = 10

K Count of array elements; dummy species index

L Step in cross section computations; dummy
species index

LL First index of species pair to be moved

LQ Location of species pair IV, JV in IQ, JQ
array

LQ1 Location in IQ, JQ array of last species pair
for step L-1 of computations

LQ2 Location in IQ, JQ array of last species pair
for step L of computations

LV Index of array elements to be moved

MM Location in V array of last parameter for
step L-1 of computations

MMV Total number of parameters in V array

MV Index for parameters in V array

NSV Total number of steps in edited cross section
computations

Card XSE 213 to Statement 300

II First index of unused storage location
for species pair

IV First index of a species pair

J Second index of species pair to be
moved

JJ Second index of unused storage location
for species pair

JV Second index of a species pair

L Step of cross section computations

LL First index of species pair to be
moved

LQ Index of species pair in IQ, JQ array

LQ2 Total number of species pairs in IQ, JQ
array

MV Dummy argument, value not required

Statement 380 to end of subroutine

I(K) Species for which cross section data are
to be supplied (common variable)

IE For IE = 1, cross section data are not
supplied for electrons

II	Absolute value of IZ for a species
IZ(L)	Charge of species L in units of electron charge (Note that IZ is negative for positively charged species)
J	Index indicating charge for species for which cross sections are to be supplied (see Table III)
JJ	Index indicating whether cross sections are to be supplied for electrons (see Table III)
KK	Index indicating charge of species for which cross sections are to be supplied (see Table III)
L	Step in cross section computations; species index
LL	Computational step at which additional cross section data are supplied for neutral-neutral interactions
NEWMV	Total number of parameters in V array
NN	Number of species for which cross section data are to be supplied

VCOUL(M, KK, J, JJ) Parameters V(M) for effective Coulomb cross sections between two species with the electronic charges KK and J, from eqs. I(100a). The parameters for JJ = 1 are for ion-ion collisions, and for JJ = 2 for ion-electron collisions

5. REFERENCES TO COMMON VARIABLES

This section presents a computer-generated analysis of references to common variables and common-equivalent variables in the NATA code. The variables are listed in alphanumeric order. For each variable, the name of the common block containing it is given. The block name / / refers to unlabelled common. An asterisk following the block name indicates that the variable does not appear in common, but is equivalenced to a common variable in the designated block in one or more routines. The routines in which each variable is referenced are indicated by numbers, based on the list of routine names given on the first page. Routine number 2 is the block data routine containing species, reaction, geometry, and other data; number 3 is the transport cross section block data routine. The routines EXP and RESET are missing from the list because the analysis was carried out for the UNIVAC version of NATA, which does not contain these routines. Neither EXP nor RESET contains any common blocks.

Lines containing a blank variable name are continuations of the preceding lines, in cases in which the number of referencing routines is too large to be printed on a single line.

Appearance of a variable in a common or equivalence statement is not considered to constitute a reference. A routine is listed as referencing a given variable if the variable appears in an arithmetic or DATA statement, a READ or WRITE statement or a NAMELIST statement, or is used as an index or a subroutine or function argument.

In the case of a common-equivalent variable, the program used to prepare the list checks for the presence of the associated common block in the routine, but does not verify the presence of the equivalence statement in the routine. For this reason, some of the listed references to common-equivalent variables are spurious. For example, in subroutine BLAYER, the variable X is equivalenced to the

variable CX in unlabelled common. Subroutines GEØMAR, ØUT1, and TRANSP are also listed as referencing X. All three of these references are spurious because X is not equivalenced to a common variable in these three routines.

In some cases, different names are used for the same common variable in different routines. All such names are included in the list below, each with its own list of referencing routines.

INDICES OF ROUTINES

1	MAIN	36	NRMAX
2	BLOCKD	37	OUT
3	BLOCKD	38	DUT1
4	AGSOLN	39	PERT
5	AXFIT	40	PIOMEG
6	AXSFCT	41	PRQP
7	RLAYER	42	PRTA
8	RLCALL	43	PUTGIN
9	BXSFCT	44	OCJUL
10	COMM	45	OFX
11	CXSFCT	46	QEXP
12	DEPIVS	47	QINTRP
13	DSMSOL	48	OLJ
14	DUMPEX	49	OMIX
15	ELCOND	50	QREPP
16	ELTIME	51	OSAME
17	EPART	52	OTAB
18	FOCALC	53	Q11
19	EXACT	54	Q12
20	FINDX	55	Q13
21	FR07FQ	56	Q14
22	GEOM	57	RADIUS
23	GEOMAR	58	READ
24	INIT	59	RESTMP
25	KANDMU	60	RNKT
26	LIST	61	SHOCK
27	MATIMV	62	SIMQ
28	MODEL	63	STUNTS
29	NEWTRAP	64	THROAT
30	NEXTMP	65	TRANSP
31	NONEC	66	WEDGE
32		67	WFSOLN
33		68	XSFCT
34		69	
35			

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VARIABLE COMMON REFERENCING ROUTINES

CF	/CONVRT/	1, 16, 32, 38, 42	1, 16, 32, 38, 41, 59, 60,
CGI		11, 16, 20, 26, 33, 66,	
CGMU		16, 26, 33,	
CGMW	/COLDSP/	1, 8, 59,	
CH		1, 11, 16, 23, 26, 32, 33, 35, 38, 41,	
CHA		1, 8, 11, 16, 23, 26, 32, 33, 35, 38, 41, 59, 60,	
CHANAM	/RDOUT	38, 58,	
CHB	/TNERK	35, 60, 38, 42,	
CHI		11, 16, 33,	
CHI1		16, 20, 33,	
CLNPI		11, 16, 64,	
CLNT		13, 16, 64,	
CLNTD	/TNDC	11, 13, 16,	
CLNIMC		11, 16,	
CM		1, 8, 9, 11, 16, 23, 26, 32, 33, 38, 39,	
CMA		1, 8, 11, 16, 21, 23, 26, 32, 33, 35, 38, 39, 41,	
CMA		59, 63,	
CMW		16, 26, 30, 58,	
CRUPLD	/BI NE	58, 16,	
CP	/CHAN *	58,	
CPWALL	/DL	8, 16, 32, 39, 67,	
CP1	/CHAN	1,	
CP2	/CHAN	2,	
CP3	/CHAN		
CP4	/CHAN		
CP5	/CHAN		
CRA			
CRP			
CRFB			
CRS			
CSTA			
CT			
CTAP			
CTAPI	/READAT	1, 11, 16, 26, 27, 32, 63,	
CTB		1, 8, 11, 16, 21, 27, 32, 35, 38, 59, 63,	
CIC		16, 23, 26, 33,	
CTD		11, 16, 35,	
CTF	/TNDC	9, 11, 13, 16, 20, 21, 23, 27, 32, 33,	
CTF3	/TONEO	35, 36, 38, 39, 41, 42, 60, 63, 64, 66,	
CTMAX	/TNERK	1, 8, 11, 16, 20, 21, 23, 26, 27, 32, 35, 38, 39,	
CTMXX		58, 59, 63, 64, 66,	
CTMXXI	/MASSFL	58,	
CTP		16, 35, 60,	
CTPL		16,	
CTT		11, 13, 16,	
CX		11, 13, 16, 21, 27, 35, 38, 42, 60,	
CXB		11, 13, 16, 21, 27, 35, 38, 42, 60,	
CXMAX		16, 35, 60,	
CXMAXI		1, 16, 6,	
CZ	/READAT	8, 16, 35, 60,	
DATAPE	/TAPOUT	16, 23, 30, 35, 58,	
		1, 2, 16, 32, 38, 58,	

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INDEXABLE COMMON REFERENCING ROUTINES

ROUTINE	COMMON	REFERENCING ROUTINES
EPI	/ELEM	2.
EP10	/ELEM	
EP2	/ELEM	2.
EP3	/ELEM	2.
EP4	/ELEM	2.
EP5	/ELEM	2.
EP6	/ELEM	
EP7	/ELEM	
EP8	/ELEM	
EP9	/ELEM	
ERR	/ERROR	1. 8. 11. 13. 15. 20. 21. 22. 23. 24. 26. 32. 34. 35. 36. 38. 39. 42. 59. 60. 65. 66. 69.
ETAJ	/ERROR	11. 16. 30. 58.
ETAJ	/ERROR	16. 27. 30. 58. 63. 64.
FACMP	/NEWMP	
FACNAM	/RDOUT	38. 58.
FAILED	/CERROR	11. 13. 16. 35. 60.
FLEWIS	/TRPROR	1. 16. 32. 66.
FLEWR	/RESRPR	
FLOW	/RDOUT	1. 38. 58.
FLUX	/RDOUT	1. 16. 23. 33. 35. 36. 41.
FSTAG	/MODPAR	
FVOUT	/OUTPUT	2. 16. 30. 32. 58.
GAMIN	/RENE	38. 47. 58.
GAMMA	/OUTPUT*	2. 35. 58.
GFLJ	/OUTPUT*	67. 58. 64.
GJ	/	1. 8. 9. 11. 21. 23. 32. 33. 35. 36. 38. 39. 60.
GJA	/	1. 16. 23. 26. 35. 36. 41. 63.
GJE	/	16. 35. 60.
GJMF	/OUTPUT	
GMA	/OUTPUT*	67.
GPRP	/OUTPUT*	58.
GP1	/MIXT	2.
GP2	/MIXT	2.
GP3	/MIXT	2.
GP4	/MIXT	2.
GP5	/MIXT	2.
GP6	/MIXT	2.
GSO	/MIXT	2.
GTFST	/RENE	35. 58.
H	/	2. 35. 58.
HDFLX	/	8. 38. 59.
HR	/BLOUT	16. 35. 60.
HS	/RDMAIN	30. 32. 35. 38. 58. 63.
HSTAG	/READAT	2. 8. 16. 38.
HTEST	/RENE	32. 38. 58. 59.
IAMBIP	/TRANS2	58. 35. 58.
IATCM	/RDMSD	7. 18. 63. 69.
IC	/LN	20. 1. 56.
		69. 16. 32. 33. 38. 58.

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VARIABLE	COMMON	REFERENCING ROUTINES
ICASE	/READAT/	2: 58:
ICCH	/TNOEO/	11: 16: 58:
ICCHAN	/READAT/	2: 58:
ICHAND	/RDLIST/	2: 58:
ICCI	/READAT/	58: 69:
ID1	/TRANS5/	66: 69:
ID2	/TRANS5/	66: 69:
IF	/READAT/	58:
IFEEP	/READAT/	58:
IELEC	/TRANS5/	17: 69:
IFAIL	/TNERK/	35: 60:
IFLOW	/TAPOUT/	
IGAS	/RDLIST/	2: 58:
IGASO	/RDLIST/	2: 58:
IGJ	/	16: 27: 30: 58: 63: 64:
IGM	/	16: 27: 30: 58: 64:
II111	/TRANS2/	
II11	/TRANI/	63: 69:
II10	/TRANI/	3: 69:
II100	/TRANI/	3:
II111	/TRANI/	3:
II112	/TRANI/	3:
II113	/TRANI/	3:
II114	/TRANI/	3:
II115	/TRANI/	3:
II116	/TRANI/	3:
II117	/TRANI/	3:
II118	/TRANI/	3:
II119	/TRANI/	3:
II12	/TRANI/	3:
II120	/TRANI/	3:
II121	/TRANI/	3:
II122	/TRANI/	3:
II123	/TRANI/	3:
II124	/TRANI/	3:
II125	/TRANI/	3:
II126	/TRANI/	3:
II127	/TRANI/	3:
II128	/TRANI/	3:
II129	/TRANI/	3:
II13	/TRANI/	3:
II130	/TRANI/	3:
II131	/TRANI/	3:
II132	/TRANI/	3:
II133	/TRANI/	3:
II134	/TRANI/	3:
II135	/TRANI/	3:
II136	/TRANI/	3:
II137	/TRANI/	3:

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TABLE REFERENCING ROUTINE

TABLE	COMMON	REFERENCING ROUTINE
1138	/TRANI	3.
1139	/TRANI	3.
1140	/TRANI	3.
1141	/TRANI	3.
1142	/TRANI	3.
1143	/TRANI	3.
1144	/TRANI	3.
1145	/TRANI	3.
1146	/TRANI	3.
1147	/TRANI	3.
1148	/TRANI	3.
1149	/TRANI	3.
1150	/TRANI	3.
1151	/TRANI	3.
1152	/TRANI	3.
1153	/TRANI	3.
1154	/TRANI	3.
1155	/TRANI	3.
1156	/TRANI	3.
1157	/TRANI	3.
1158	/TRANI	3.
1159	/TRANI	3.
1160	/TRANI	3.
1161	/TRANI	3.
1162	/TRANI	3.
1163	/TRANI	3.
1164	/TRANI	3.
1165	/TRANI	3.
1166	/TRANI	3.
1167	/TRANI	3.
1168	/TRANI	3.
1169	/TRANI	3.
1170	/TRANI	3.
1171	/TRANI	3.
1172	/TRANI	3.
1173	/TRANI	3.
1174	/TRANI	3.
1175	/TRANI	3.
1176	/TRANI	3.
1177	/TRANI	3.
1178	/TRANI	3.
1179	/TRANI	3.
1180	/TRANI	3.
1181	/TRANI	3.
1182	/TRANI	3.

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TABLE COMMON REFERENCING ROUTINES

TABLE	COMMON	REFERENCING ROUTINES
ISSP3	/	16. 21. 35. 39. 60.
ISSP4	/	16. 21. 35. 60.
ISSUPR	/	35.
ISW1A	/	1. 16. 58.
ISW1B	/	1. 16. 58. 63. 66.
ISW2A	/	1. 16. 58. 59. 59.
ISW2B	/	1. 16. 58.
ISW3A	/	1. 16. 9. 13. 16. 23. 24. 30. 35. 38. 42. 58. 65.
ISW3B	/	1. 16. 58.
ISW4A	/	1. 16. 16. 58.
ISW4B	/	1. 8. 16. 58.
ISW5A	/	1. 16. 58. 59. 35. 42. 58. 60.
ISW5B	/	1. 11. 16. 58.
ISW6A	/	1. 16. 58.
ISW6B	/	1. 16. 38. 58.
ISW7B	/	2. 58. 47. 58. 66.
ISW9B	/	3. 43. 67.
ISW9B	/	2. 58. 67.
ITB	/	16. 35.
ITPOUT	/	1. 2. 16. 32. 38. 58.
ITR	/	16. 19. 30. 58.
ITYPER	/	1. 32. 78.
IUPD	/	R. 16. 21. 35. 42. 58. 65.
IZERO	/	R. 16. 30. 58. 64.
I1	/	1. 58.
I1C	/	58.
I1CO	/	58.
I11	/	58.
I12	/	58.
I13	/	58.
I14	/	58.
I15	/	58.
I16	/	58.
I17	/	58.
I1P	/	58.
I19	/	58.
I20	/	58.
I21	/	58.
I22	/	58.
I23	/	58.
I24	/	58.
I25	/	58.
I26	/	58.
I27	/	58.
I28	/	58.
I29	/	58.
I3	/	58.
I30	/	58.
I31	/	58.

REFERENCING ROUTINES

COMMON

VARIABLE

I32	/	58.
I33	/	58.
I34	/	58.
I35	/	58.
I36	/	58.
I37	/	58.
I38	/	58.
I39	/	58.
I4	/	58.
I40	/	58.
I41	/	58.
I42	/	58.
I43	/	58.
I44	/	58.
I45	/	58.
I46	/	58.
I47	/	58.
I48	/	58.
I49	/	58.
I5	/	58.
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I51	/	58.
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I53	/	58.
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I55	/	58.
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I60	/	58.
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JJ13	/TRANJ	3.
JJ14	/TRANJ	3.
JJ15	/TRANJ	3.
JJ16	/TRANJ	3.
JJ17	/TRANJ	3.
JJ18	/TRANJ	3.
JJ19	/TRANJ	3.
JJ2	/TRANJ	3.
JJ20	/TRANJ	3.
JJ21	/TRANJ	3.
JJ22	/TRANJ	3.
JJ23	/TRANJ	3.
JJ24	/TRANJ	3.
JJ25	/TRANJ	3.
JJ26	/TRANJ	3.
JJ27	/TRANJ	3.

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JJ29	/TRANJ /	3.
JJ3	/TRANJ /	3.
JJ30	/TRANJ /	3.
JJ31	/TRANJ /	3.
JJ32	/TRANJ /	3.
JJ33	/TRANJ /	3.
JJ34	/TRANJ /	3.
JJ35	/TRANJ /	3.
JJ36	/TRANJ /	3.
JJ37	/TRANJ /	3.
JJ38	/TRANJ /	3.
JJ39	/TRANJ /	3.
JJ4	/TRANJ /	3.
JJ40	/TRANJ /	3.
JJ41	/TRANJ /	3.
JJ42	/TRANJ /	3.
JJ43	/TRANJ /	3.
JJ44	/TRANJ /	3.
JJ45	/TRANJ /	3.
JJ46	/TRANJ /	3.
JJ47	/TRANJ /	3.
JJ48	/TRANJ /	3.
JJ49	/TRANJ /	3.
JJ5	/TRANJ /	3.
JJ50	/TRANJ /	3.
JJ51	/TRANJ /	3.
JJ52	/TRANJ /	3.
JJ53	/TRANJ /	3.
JJ54	/TRANJ /	3.
JJ55	/TRANJ /	3.
JJ56	/TRANJ /	3.
JJ57	/TRANJ /	3.
JJ58	/TRANJ /	3.
JJ59	/TRANJ /	3.
JJ6	/TRANJ /	3.
JJ60	/TRANJ /	3.
JJ61	/TRANJ /	3.
JJ62	/TRANJ /	3.
JJ63	/TRANJ /	3.
JJ64	/TRANJ /	3.
JJ65	/TRANJ /	3.
JJ66	/TRANJ /	3.
JJ67	/TRANJ /	3.
JJ68	/TRANJ /	3.
JJ69	/TRANJ /	3.
JJ7	/TRANJ /	3.
JJ70	/TRANJ /	3.
JJ71	/TRANJ /	3.
JJ72	/TRANJ /	3.

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JJ75	/TRANJ /	
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JJ82	/TRANJ /	
JJ83	/TRANJ /	
JJ84	/TRANJ /	
JJ85	/TRANJ /	
JJ86	/TRANJ /	
JJ87	/TRANJ /	
JJ88	/TRANJ /	
JJ89	/TRANJ /	
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JJ92	/TRANJ /	
JJ93	/TRANJ /	
JJ94	/TRANJ /	
JJ95	/TRANJ /	
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J11	/TRANJ /	58.
J12	/TRANJ /	58.
J13	/TRANJ /	58.
J14	/TRANJ /	58.
J15	/TRANJ /	58.
J16	/TRANJ /	58.
J17	/TRANJ /	58.
J18	/TRANJ /	58.
J19	/TRANJ /	58.
J2	/TRANJ /	58.
J20	/TRANJ /	58.
J21	/TRANJ /	58.
J22	/TRANJ /	58.
J23	/TRANJ /	58.
J24	/TRANJ /	58.
J25	/TRANJ /	58.

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J26	/TRANJ	58.
J27	/TRANJ	58.
J28	/TRANJ	58.
J29	/TRANJ	58.
J30	/TRANJ	58.
J31	/TRANJ	58.
J32	/TRANJ	58.
J33	/TRANJ	58.
J34	/TRANJ	58.
J35	/TRANJ	58.
J36	/TRANJ	58.
J37	/TRANJ	58.
J38	/TRANJ	58.
J39	/TRANJ	58.
J40	/TRANJ	58.
J41	/TRANJ	58.
J42	/TRANJ	58.
J43	/TRANJ	58.
J44	/TRANJ	58.
J45	/TRANJ	58.
J46	/TRANJ	58.
J47	/TRANJ	58.
J48	/TRANJ	58.
J49	/TRANJ	58.
J50	/TRANJ	58.
J51	/TRANJ	58.
J52	/TRANJ	58.
J53	/TRANJ	58.
J54	/TRANJ	58.
J55	/TRANJ	58.
J56	/TRANJ	58.
J57	/TRANJ	58.
J58	/TRANJ	58.
J59	/TRANJ	58.
J60	/TRANJ	58.
J61	/TRANJ	58.
J62	/TRANJ	58.
J63	/TRANJ	58.
J64	/TRANJ	58.
J65	/TRANJ	58.
J66	/TRANJ	58.
J67	/TRANJ	58.
J68	/TRANJ	58.
J69	/TRANJ	58.
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RP12	/REAC	2, 58,
RP13	/REAC	2, 58,
RP14	/REAC	2, 58,
RP15	/REAC	2, 58,
RP16	/REAC	2, 58,
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RP35	/	/REAC	2. 58.
RP36	/	/REAC	2. 58.
RP37	/	/REAC	2. 58.
RP38	/	/REAC	2. 58.
RP39	/	/REAC	2. 58.
RP40	/	/REAC	2. 58.
RP41	/	/REAC	2. 58.
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RP43	/	/REAC	2. 58.
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RP46	/	/REAC	2. 58.
RP47	/	/REAC	2. 58.
RP48	/	/REAC	2. 58.
RP49	/	/REAC	2. 58.
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RP51	/	/REAC	2. 58.
RP52	/	/REAC	2. 58.
RP53	/	/REAC	2. 58.
RP54	/	/REAC	2. 58.
RP55	/	/REAC	2. 58.
RP56	/	/REAC	2. 58.
RP57	/	/REAC	2. 58.
RP58	/	/REAC	2. 58.
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RP60	/	/REAC	2. 58.
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RP62	/	/REAC	2. 58.
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RP78	/REAC	2. 58.
RP79	/REAC	2. 58.
RP8	/REAC	2. 58.
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RP85	/REAC	2. 58.
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SP21	/	2, 58,
SP22	/	2, 58,
SP23	/	2, 58,
SP24	/	2, 58,
SP25	/	2, 58,
SP26	/	2, 58,
SP27	/	2, 58,
SP28	/	2, 58,
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TL12	/TRANS4/	3,
TL13	/TRANS4/	3,
TL14	/TRANS4/	3,
TL15	/TRANS4/	3,
TL16	/TRANS4/	3,
TL17	/TRANS4/	3,
TL18	/TRANS4/	3,
TL19	/TRANS4/	3,
TL2	/TRANS4/	3,
TL20	/TRANS4/	3,
TL21	/TRANS4/	3,
TL22	/TRANS4/	3,
TL23	/TRANS4/	3,
TL24	/TRANS4/	3,
TL25	/TRANS4/	3,
TL26	/TRANS4/	3,
TL27	/TRANS4/	3,
TL28	/TRANS4/	3,
TL29	/TRANS4/	3,
TL7	/TRANS4/	3,
TL30	/TRANS4/	3,
TL31	/TRANS4/	3,
TL32	/TRANS4/	3,
TL33	/TRANS4/	3,
TL34	/TRANS4/	3,
TL35	/TRANS4/	3,
TL36	/TRANS4/	3,
TL37	/TRANS4/	3,
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VARIABLE

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VV10	/TRANV /	3.
VV100	/TRANV /	3.
VV11	/TRANV /	3.
VV12	/TRANV /	3.
VV13	/TRANV /	3.
VV14	/TRANV /	3.
VV15	/TRANV /	3.
VV16	/TRANV /	3.
VV17	/TRANV /	3.
VV18	/TRANV /	3.
VV19	/TRANV /	3.

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VARIABLE

VARIABLE	COMMON	REFERENCING ROUTINES
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VV20	/TRANV /	3.
VV21	/TRANV /	3.
VV22	/TRANV /	3.
VV23	/TRANV /	3.
VV24	/TRANV /	3.
VV25	/TRANV /	3.
VV26	/TRANV /	3.
VV27	/TRANV /	3.
VV28	/TRANV /	3.
VV29	/TRANV /	3.
VV3	/TRANV /	3.
VV10	/TRANV /	3.
VV31	/TRANV /	3.
VV32	/TRANV /	3.
VV33	/TRANV /	3.
VV34	/TRANV /	3.
VV35	/TRANV /	3.
VV36	/TRANV /	3.
VV37	/TRANV /	3.
VV38	/TRANV /	3.
VV39	/TRANV /	3.
VV4	/TRANV /	3.
VV40	/TRANV /	3.
VV41	/TRANV /	3.
VV42	/TRANV /	3.
VV43	/TRANV /	3.
VV44	/TRANV /	3.
VV45	/TRANV /	3.
VV46	/TRANV /	3.
VV47	/TRANV /	3.
VV48	/TRANV /	3.
VV49	/TRANV /	3.
VV5	/TRANV /	3.
VV50	/TRANV /	3.
VV51	/TRANV /	3.
VV52	/TRANV /	3.
VV53	/TRANV /	3.
VV54	/TRANV /	3.
VV55	/TRANV /	3.
VV56	/TRANV /	3.
VV57	/TRANV /	3.
VV58	/TRANV /	3.
VV59	/TRANV /	3.
VV6	/TRANV /	3.
VV60	/TRANV /	3.
VV61	/TRANV /	3.
VV62	/TRANV /	3.
VV63	/TRANV /	3.
VV64	/TRANV /	3.

REFERENCING ROUTINES

COMMON

RIABLE

VV65	/TRANV /	
VV66	/TRANV /	
VV67	/TRANV /	
VV68	/TRANV /	
VV69	/TRANV /	
VV7	/TRANV /	
VV70	/TRANV /	
VV71	/TRANV /	
VV72	/TRANV /	
VV73	/TRANV /	
VV74	/TRANV /	
VV75	/TRANV /	
VV76	/TRANV /	
VV77	/TRANV /	
VV78	/TRANV /	
VV79	/TRANV /	
VV8	/TRANV /	
VV80	/TRANV /	
VV81	/TRANV /	
VV82	/TRANV /	
VV83	/TRANV /	
VV84	/TRANV /	
VV85	/TRANV /	
VV86	/TRANV /	
VV87	/TRANV /	
VV88	/TRANV /	
VV89	/TRANV /	
VV9	/TRANV /	
VV90	/TRANV /	
VV91	/TRANV /	
VV92	/TRANV /	
VV93	/TRANV /	
VV94	/TRANV /	
VV95	/TRANV /	
VV96	/TRANV /	
VV97	/TRANV /	
VV98	/TRANV /	
VV99	/TRANV /	
V1	/TRANV /	58.
V10	/TRANV /	58.
V100	/TRANV /	58.
V11	/TRANV /	58.
V12	/TRANV /	58.
V13	/TRANV /	58.
V14	/TRANV /	58.
V15	/TRANV /	58.
V16	/TRANV /	58.
V17	/TRANV /	58.
V18	/TRANV /	58.
V19	/TRANV /	58.

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REFERENCING ROUTINES

COMMON

VARIABLE

V2	/TRANV /	58.
V20	/TRANV /	58.
V21	/TRANV /	58.
V22	/TRANV /	58.
V23	/TRANV /	58.
V24	/TRANV /	58.
V25	/TRANV /	58.
V26	/TRANV /	58.
V27	/TRANV /	58.
V28	/TRANV /	58.
V29	/TRANV /	58.
V3	/TRANV /	58.
V30	/TRANV /	58.
V31	/TRANV /	58.
V32	/TRANV /	58.
V33	/TRANV /	58.
V34	/TRANV /	58.
V35	/TRANV /	58.
V36	/TRANV /	58.
V37	/TRANV /	58.
V38	/TRANV /	58.
V39	/TRANV /	58.
V4	/TRANV /	58.
V40	/TRANV /	58.
V41	/TRANV /	58.
V42	/TRANV /	58.
V43	/TRANV /	58.
V44	/TRANV /	58.
V45	/TRANV /	58.
V46	/TRANV /	58.
V47	/TRANV /	58.
V48	/TRANV /	58.
V49	/TRANV /	58.
V5	/TRANV /	58.
V50	/TRANV /	58.
V51	/TRANV /	58.
V52	/TRANV /	58.
V53	/TRANV /	58.
V54	/TRANV /	58.
V55	/TRANV /	58.
V56	/TRANV /	58.
V57	/TRANV /	58.
V58	/TRANV /	58.
V59	/TRANV /	58.
V6	/TRANV /	58.
V60	/TRANV /	58.
V61	/TRANV /	58.
V62	/TRANV /	58.
V63	/TRANV /	58.
V64	/TRANV /	58.

REFERENCING ROUTINES

VARIABLE	COMMON	REFERENCING ROUTINES
V65	/TRANV /	58.
V66	/TRANV /	58.
V67	/TRANV /	58.
V68	/TRANV /	58.
V69	/TRANV /	58.
V70	/TRANV /	58.
V71	/TRANV /	58.
V72	/TRANV /	58.
V73	/TRANV /	58.
V74	/TRANV /	58.
V75	/TRANV /	58.
V76	/TRANV /	58.
V77	/TRANV /	58.
V78	/TRANV /	58.
V79	/TRANV /	58.
V80	/TRANV /	58.
V81	/TRANV /	58.
V82	/TRANV /	58.
V83	/TRANV /	58.
V84	/TRANV /	58.
V85	/TRANV /	58.
V86	/TRANV /	58.
V87	/TRANV /	58.
V88	/TRANV /	58.
V89	/TRANV /	58.
V90	/TRANV /	58.
V91	/TRANV /	58.
V92	/TRANV /	58.
V93	/TRANV /	58.
V94	/TRANV /	58.
V95	/TRANV /	58.
V96	/TRANV /	58.
V97	/TRANV /	58.
V98	/TRANV /	58.
V99	/TRANV /	58.
WEDGEM	/SWITCH/	16. 23. 35.
WX	/RDWEDG/	30. 32. 58. 67.
WCAVE	/AVG	2. 23. 35. 58.
WXI	/RDWEDG/	2. 30. 58. 67.
WXI	/RDWEDG/	2. 30. 58. 67.
XI	/BLNE /	8. 25. 38. 66.
XMJAT	/TNDG /	13. 16. 20. 33. 63. 64.
XMJATD	/MODPAR/	11. 13. 15. 16.
XMOOP1	/MODPAR/	2. 16. 30. 58.
XMPI	/MODPAR/	16. 23. 35. 58.

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VARIABLE	COMMON	REFERENCING ROUTINES
XNUI	/	11, 16, 35, 58
XNUIJ	/	11, 30, 35, 58
XNUIJP	/	11, 30, 35, 58
XP	/SWITCH/*	8, 23, 35, 58
XPB	/SWITCH/	16, 23, 35, 58
XSN	/BLOUT/	2, 8, 15, 35, 38
XX	/TAPOUT/	1, 38, 58
XZERO	/BL/	8, 16, 23, 58
XZEROI	/READAT/	58, 57
YOZO	/BLRAD/	9, 20, 25, 32
ZCAP	/EQC/	16, 20, 25, 32
ZCH	/EQC/	16, 20, 25, 32
ZCM	/EQC/	16, 20, 25, 32
ZGMU	/EQC2/	20, 26, 69
ZK2	/TRANSI/	28, 65, 41, 33
ZP	/	16, 23, 26, 33
ZPA	/	16, 23, 26, 33
ZPRP	/	58, 26
ZPZ	/NOZZ/	20, 26
ZP1	/EQC2/	2, 2
ZP10	/NOZZ/	2, 2
ZP11	/NOZZ/	2, 2
ZP12	/NOZZ/	2, 2
ZP13	/NOZZ/	2, 2
ZP14	/NOZZ/	2, 2
ZP15	/NOZZ/	2, 2
ZP16	/NOZZ/	2, 2
ZP17	/NOZZ/	2, 2
ZP18	/NOZZ/	2, 2
ZP19	/NOZZ/	2, 2
ZP2	/NOZZ/	2, 2
ZP30	/NOZZ/	2, 2
ZP3	/NOZZ/	2, 2
ZP4	/NOZZ/	2, 2
ZP5	/NOZZ/	2, 2
ZP6	/NOZZ/	2, 2
ZP7	/NOZZ/	2, 2
ZP8	/NOZZ/	2, 2
ZP9	/NOZZ/	2, 2
ZP9P	/EQC/	16, 20, 26, 32
ZRHQ	/EQC/	16, 20, 26, 32
ZSFN	/EQC/	16, 20, 26, 32

COMPLETED

6. LISTING OF THE NATA CODE

The present section consists of a complete source program listing of the NATA code. The routines appear in the same order in which their names are listed at the beginning of Section 5, i.e., main program, block data routines, and then all subroutines and functions in alphanumeric order of their names.

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C MAIN PROGRAM FOR NOZZLE FLOW SOLUTION
LOGICAL SUPGO,DATAPE,NOTRAN
LOGICAL ERR
REAL ACUM(30),ELMENT(10),HP(20)
DOUBLE PRECISION AA,AAA,CAPX,GJ,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A, AFNTS, AFNX, AMACH, AR, ARBA, ARBB, BZERO,
C, CARB, CH, CHT, CLNT, CM, CMA, CRA, CRP,
1 CRRB, CRS, CSTA, CT, CTAP, CTB, CTC, CTMAX, CTMXX,
2 CTP, CTPL, CTT, CX, CXB, CXMAX, DATESY, DBTEST, DELT1,
3 DELT2, DELTAX, DLOGA, DLOGR, DT, ENT, FLUX, HDELX, PCT,
4 DELT2, DELTAX, DLOGA, DLOGR, DT, ENT, FLUX, HDELX, PCT,
5 PCTEST, PRES, PRESA, PRESB, PRESTH, PRHG, RHAP, RHO, RHOB,
6 RHOBAR, RHOC, RHQP, RHPL, RHTH, ROBAR, ROBARP, SCRG, SDT,
7 SEN, SHPG, SHOC, SC, SL, SL64, SU, SU2, SUMG,
8 TEST, TESTB, TPRINT, TSTOP, ZP, ZPA
COMMON BE(64), BET(20), BLBK(31), CAI(64), CAPXTH(20), MAI
1 CCPJ(20), CFACT(64), CGI(20), CGMU(20), CHI(64), CHI(20), MAI
2 CLN14C(64), CLNPI(64), CMW(20), ETAI(64), FTAJ(20), GJA(20), MAI
3 GJB(20), PERTGJ(20), PGJ(20), PI(64), PICH(64), QM(20), MAI
4 GJB(20), SAJ(20), SDCHI(64), SENT(20), SHJ(20), SHJA(20), MAI
5 SKTL(20), SS(20), TB(30), TFA(20), TFC(20), TFC(20), MAI
6 TFD(20), TFE(20), TFK(20), THEV(20), XMJAT(20), XNUI(64)
COMMON B, TA(64,20), ELJ(10,20), GELJ(10,20)
1 XNUIJ(64,20), XNUIJR(64,20)
COMMON IC, INEQV, IP, IRUN, ISC, ISCP1,
1 ISMC, ISYCNR, ISR, ISS, ISSNR, ISSP1, ISSP2, ISSP3, ISSP4,
2 ISW1A, ISW1B, ISW2A, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
3 ISW5B, ISW6A, ISW6B, IUPD, IZER, JJK, LC, M1, NFIT,
4 NIT, NNN, NNS, NQT, NTEST
COMMON I, J, K(20), ITB(5), KUR(64,20), LPIJ(20,10)
COMMON ACOM, LLMENI, HP
COMMON ION AAA(22,24), BTAA(64,20), CAPQ(31), CC(20), DGJ(20), GJ(20), SBMAI
1 J(20), SDGJ(20), SHJAP(20), THEVP(20)
EQUIVALENCE (AA(1,1),AAA(1,1)),(BETA(1,1),RTA(1,1)),
1 (BLBK(1),CAPQ(1)),(CAPXTH(1),CCI(1)),(GJA(1),DGJ(1)),
2 (CAPX(1),GJ(1)),(SAJ(1),SBJ(1)),(SS(1),SDGJ(1)),
3 (SHJA(1),SHJAP(1)),(THEV(1),THEVP(1))
EQUIVALENCE (NOZZLE,NPROFL(1))
COMMON /ERROR/ ERR
COMMON /TEVPRY/ SAVEC(20)
COMMON /MASSEL/ SPASS,CTMXXI,TSTEP1,IS(20)
COMMON /BL/ DELBL(2),BLINT(2),XERO,TWALL,CPWALL,VISROT,DIAM(2),
1 SWRO, IM,IPGINT
DIMENSION GMF(20)
COMMON /RDMAIN/ HS,SUPGO,MFITER,NOTRAN
COMMON /NEQ/ GMNST(2),DDELBL(2)
COMMON /COLD/PG/ CGMW,OPJ(10),IJCS(10),NCS
COMMON /TRPROP/ VISC,FR,SIGMA,FLEWIS
COMMON /RESPRP/ VISC,PRR,SIGR,FLEWR
COMMON /TPOUT/ XX(10),TPOUT,NRCOUT,IFLOW,ITYPER,IMP,DATAPE
COMMON /AREA/ A(PI(1,1)),PARAM(3,12,2),RTHCM(2),NSECT(2),
1 NSECTU(2),ISHAPE(1,2),NPROFL(2),NPPFLS,NBL
COMMON /REL/ST/ IGAL,IGA30,NOZZO,IMAND
COMMON /CFNVRT/ CF(6)

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COMMON /OUTPUT/ FVOUT(35),GJMF(20)
COMMON /RDOU/ FLOW,FACNAM,CHANAM,LIMOU
COMMON /SS/ CAS,US
COMMON /SPEC/ SPRP(43,30)
COMMON /LN/ ISATOM,ISMOL,JATOM,JMOL
COMMON /EELEM/ EEP(2,10)
COMMON /THRT/ RSA
DATA NCASE,NERROR,NRCTOT,IRCO /4*0/, ONE /1./, PIV /3.14159/
ISW1A=1
ISW2A=1
ISW3A=1
ISW4A=0
ISW5A=0
ISW6A=0
ISW1B=0
ISW2B=0
IS3=1
ISW4B=0
ISW5B=0
ISW6B=1
IRUN=0
MFIITER=1
NOTRAN=.FALSE.
CALL ELTIME (ET0,0)
ISW3D=IS3
IS3=ISW3B
IF (.NOT.NOTRAN) GO TO 20
ISW3B=0
VISC=0.
PR=0.
SIGMA=0.
FLEWIS=0.
CALL LIST
IF (.NOTRAN) WRITE (6,330)
ET=ET0
CALL ELTIME (ET,1)
NRCOUT=0
NCASE=NCASE+1
ERR=.FALSE.
OMDST(1)=1.
OMDST(2)=1.
IF (ISW2B.GT.0) SM=0.
CALL IN GAS
IF (ERR) GO TO 290
IF (ISW2B.LE.0) CALL RESTMP
IF (ERR) GO TO 290
IF (ISW2B.LT.0) GO TO 30
CALL INIT
CALL INTA
IF (ERR) GO TO 290
IF (ISW6A.GE.0) GO TO 40
CALL STUNTS
IRCO=1
GO TO 300

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40 IF (ISW6A.EQ.2) GO TO 140
IF (SUPGO.AND.IRCO.EQ.0) GO TO 50
IF (NOTRAN) GO TO 60
CALL TRANSX
IF (ERR) GO TO 290
IRCO=0
IF (ISWIB.EQ.0) GO TO 60
CALL STUNT2
IF (ISWIB.LT.0) GO TO 140
CT=TWALL/CTAP
CALL THERM1
CPWALL=0.
DO 70 K=1,NCS
I=IJS(K)
CPWALL=CPWALL+OPJ(K)*CCPJ(I)
CPWALL=CPWALL*CRAG/CGMW
WRITE (6,360) CPWALL,TWALL
IF (NCASE.EQ.1) CALL OUT
IF (NPRFLS.EQ.2) GO TO 80
RO=RTHCM(1)
GO TO 90
80 RO=SQRT(1.27324*RTHCM(1)*RTHCM(2))
90 IF (ISW3P.EQ.0.OR.ISW2B.GT.0.OR.MFITER.EQ.0) GO TO 140
DO 100 I=1,ISS
SAVEC(I)=GJA(I)*CMA
CALL TRANSP (CTAP,PRESA)
IF (ERR) GO TO 290
VISROT=VISC/CTAP
HOPECHA*CRP/CMA
SW=CPWALL*TWALL/HOP-1.
US=SQRT(8.31436E7*CTAP/CMA)
CALL EQUIL (1)
IF (NPRFLS.EQ.1) GO TO 110
ATH=4.*RTHCM(1)-RO*DELBL(1))*RTHCM(2))-RO*DELBL(2))
GO TO 130
110 IF (JDIM.EQ.0) GO TO 120
ATH=PIV*(RO*OMDST(1))*2
GO TO 130
120 ATH=2.*RO*OMDST(1)
130 SMASS=453.59*FLOW/ATH
CALL PESTMP
IF (ISW2B.LT.0) GO TO 140
CALL INIT
CALL INTA
CF(1)=RHAP*(30.48)**3/453.59
US=SQRT(8.31436E7*CTAP/CMA)
CF(2)=US/30.48
CF(3)=30.48/453.59
CF(4)=1./2.54
CF(5)=1.8*CTAP*1.98647/CMA
CF(6)=RO/2.54
DO 150 I=1,ISS
SAVEC(I)=GJA(I)*CMA
CT=1.
PRES=1.

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RHO=1.
SU=0.
CH=CHA
CM=CMA
WRITE (6,370)
CALL OUT1
IF (ERR) GO TO 290
CALL ELTIME (ET,1)
IF (ISW6A.EQ.2.OR.ISW18.LT.0) GO TO 300
IF (NOTRAN) GO TO 160
CALL TRANSP (CTAP,PRESA)
IF (ERR) GO TO 290
VISC=VISC*CF(3)
WRITE (6,380) VISC,PR,SIGMA,FLEWIS
VISC=VISC
PR=PR
SIGM=SIGMA
FLEWR=FLEWIS
VISROT=VISC/CTAP
HOP=CHA*CRP/CMA
SW=CPWALL*TWALL/HOP-1.
IF (ISW1A.EQ.IZERO) GO TO 190
IF (.NOT.DATAPE) GO TO 180
ISOLN=1
ITYPER=1
XXX(1)=NOZZLE+0.1
XXX(2)=FLOW
XXX(3)=PRESA
XXX(4)=CF(S)*CHA
XXX(5)=DIAM(1)
XXX(6)=CTAP
XXX(7)=CF(1)
XXX(8)=SEN
XXX(9)=CMA
XXX(10)=DIAM(2)
NRROUT=NRROUT+1
WRITE (ITPOUT) ITYPER,ISOLN,(XXX(I),I=1,10)
GO TO (180,210,280), ISOLN
CALL FRUZEN GO TO 290
IF (ERR) GO TO 290
CT=CTMAX
PRES=PRESTH
RHO=RHTH
SU=SM/RHTH
CH=CHA-0.5*SU**2
WRITE (6,390)
CALL OUT1
CALL ELTIME (ET,1)
CALL NRM MAX
IF (ERR) GO TO 290
PRESTH=PRES
UTH=SU
CHTH=CH
CMTH=CM

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DO 200 I=1,ISS
GMF(I)=GJ(I)*CM
IF (ISW3A.EQ.0) GO TO 220
ISOLN=2
IF (DATAPE) GO TO 170
CALL EQUIL (2)
IF (ERR) GO TO 290
IM=1
JJK=0
ISMCNR=ISMC
ISSNR=ISS
WRITE (6,400)
CT=CTMAX
PRES=PRESTH
RHO=RHTH
SU=UTH
CM=CMTH
DO 230 I=1,ISS
SAVEC(I)=GMF(I)
CALL OUT1
IF (ERR) GO TO 290
CALL ELTIME (ET,1)
IF (ISW2A.EQ.IZERO) GO TO 300
DO 240 I=1,ISS
GJ(I)=GJA(I)
PRES=1.0
N=0
SUM1=ALOG(RHTH)
A=4.+8.*SUM1
N=N+1
IF (N-50) 260,270,270
SUM2=A+2.
SUM3=RHTH**A
D=SUM3*SUM2-2.
D1=SUM3*(SUM2*SUM1+1.)
ATEST=A
A=A-D/D1
IF (ABS(D).GT..00001) GO TO 250
IF (ABS(ATEST-A).GT..00001) GO TO 250
C=(RHTH**2)*(1.-RHTH**A)
DELT=DELT1
IF (ISW3B.NE.0) DELT=0.049*(1.-CTMAX)
CT=1.0-DELT
CM=CMA
CALL NEWRAP
IF (ERR) GO TO 290
AFNTS=SM/FLUX
CALL FINDX (AFNTS,-ONE,CX)
WRITE (6,340) A,C
ISOLN=3
IF (DATAPE) GO TO 170
CALL NONEQ
CALL ELTIME (ET,1)
IF (.NOT.ERR) GO TO 300
  
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29 CALL DUMPEX
300 NRCCTOT=NRCCTOT+1
WRITE (6,420) IRUN,NCASE,NRCOUT,ITPOUT,NRCCTOT
ELTMIN=(ET-ET0)/60.
IF (ISW4A) 10,310,10
NCOMPL=NCASE-NERROR
WRITE (6,350) NCOMPL,NERROR
IF (.NOT.DATAPE) GO TO 320
REWIND ITPOUT
CALL EXIT

320
C
C
C
330 FORMAT (43H0TRANSPORT PROPERTY CALCULATIONS SUPPRESSED)
340 FORMAT (20H0 DENSITY FIT-ALPHA=.1PE15.7,10H CONSTANT=.E15.7)
350 FORMAT (//15,16H CASES COMPLETED/15,13H CASES FAILED)
360 FORMAT (27H0SPECIFIC HEAT OF COLD GAS=.F7.4,16H BTU/LB-DEG R AT.F8MAI 295
1.2,6H DEG K)
370 FORMAT (1H1,9X,24H- RESERVOIR CONDITIONS -)
380 FORMAT (//31H RESERVOIR TRANSPORT PROPERTIES//19H VISCOSITY
1=.3X,1PE9.2,11H LBM/FT-SEC/19H PRANDTL NUMBER =.0PF12.5/19H SI MAI 299
=.3X,1PE9.2,7H MHO/CM/19H LEWIS NUMBER =.0PF10. MAI 300
33)
390 FORMAT (1H1,7X,28H- FROZEN THROAT CONDITIONS -)
400 FORMAT (1H1,6X,33H- EQUILIBRIUM THROAT CONDITIONS -)
410 FORMAT (24H CASE EXECUTION TIME WAS.F6.2,8H MINUTES)
420 FORMAT (8HORUN NO.,17.5X,4HCASE,14,23H OF THIS JOB COMPLETED.,1S,2MAI 305
14H RECORDS WRITTEN ON TAPE.13,1H.,17,36H RECORDS WRITTEN SO FAR IN MAI 306
2 THIS JOB.)
END
MAI 307
MAI 308-

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BLOCK DATA
LOGICAL DATAPE, WEDGEM, AXISYM, READXS, REI, AAMS, AXIMOD
DOUBLE PRECISION AIN
COMMON /ELEM/ EP1(2), EP2(2), EP3(2), EP4(2), EP5(2), EP6(2), EP7(2),
1 EP8(2), EP9(2), EP10(2)
1 COMMON /SPEC/ SP1(43), SP2(43), SP3(43), SP4(43), SP5(43), SP6(43),
SP7(43), SP8(43), SP9(43), SP10(43), SP11(43), SP12(43), SP13(43),
2 SP14(43), SP15(43), SP16(43), SP17(43), SP18(43), SP19(43),
3 SP20(43), SP21(43), SP22(43), SP23(43), SP24(43), SP25(43),
4 SP26(43), SP27(43), SP28(43), SP29(43), SP30(43)
COMMON /REAC/ RP1(29), RP2(29), RP3(29), RP4(29), RP5(29), RP6(29),
1 RP7(29), RP8(29), RP9(29), RP10(29), RP11(29), RP12(29), RP13(29),
2 RP14(29), RP15(29), RP16(29), RP17(29), RP18(29), RP19(29),
3 RP20(29), RP21(29), RP22(29), RP23(29), RP24(29), RP25(29),
4 RP26(29), RP27(29), RP28(29), RP29(29), RP30(29), RP31(29),
5 RP32(29), RP33(29), RP34(29), RP35(29), RP36(29), RP37(29),
6 RP38(29), RP39(29), RP40(29), RP41(29), RP42(29), RP43(29),
7 RP44(29), RP45(29), RP46(29), RP47(29), RP48(29), RP49(29),
8 RP50(29), RP51(29), RP52(29), RP53(29), RP54(29), RP55(29),
9 RP56(29), RP57(29), RP58(29), RP59(29), RP60(29), RP61(29),
A RP62(29), RP63(29), RP64(29), RP65(29), RP66(29), RP67(29),
B RP68(29), RP69(29), RP70(29), RP71(29), RP72(29), RP73(29),
C RP74(29), RP75(29), RP76(29), RP77(29), RP78(29), RP79(29),
D RP80(29), RP81(29), RP82(29), RP83(29), RP84(29), RP85(29),
E RP86(29), RP87(29), RP88(29), RP89(29), RP90(29), RP91(29),
F RP92(29)
COMMON /MIXT/ GP1(124), GP2(124), GP3(124), GP4(124), GP5(124),
1 GP6(124)
COMMON /NOZZ/ ZP1(64), ZP2(64), ZP3(64), ZP4(64), ZP5(64), ZP6(64),
1 ZP7(64), ZP8(64), ZP9(64), ZP10(64), ZP11(64), ZP12(64), ZP13(64),
2 ZP14(64), ZP15(64), ZP16(64), ZP17(64), ZP18(64), ZP19(64),
3 ZP20(64)
COMMON /AREA/ ATPI(11,2), PARAM(3,12,2), RTHCM(2), NSECT(2),
1 NSECTU(2), ISHAPE(12,2), NPAC, CIL(2), NPRFLS, NBL
COMMON /MASSFL/ SMASS, CTMXXI, TSTUPI, IS(20)
COMMON /BL/ DELBL(2), BLINT(2), XZERO, TWALL, CPWALL, VISROT, DIAM(2),
1 SW, RO, JOIM, IPCINT
COMMON /MODPAR/ XMP1, DXMP, FSTAG, CATFAC, TMODEL, XMODP1, DXMODP,
1 TPLATE, KDIM
COMMON /RDLIST/ IGAS, IGASO, NOZZO, ICHANO
COMMON /MODPT/ TSDIAM(20), TSAR(20), NTS, MBL
COMMON /TAPOUT/ XXX(10), ITPOUT, NRCOUT, IFLOW, ITPER, IMP, DATAPE
COMMON /BLOW/ REPF, THE TA(2), SN(2), XSN(2), PRREF, HR, QWDOT(2),
1 TAUW(2)
COMMON /NEWMP/ FACMP, NMODPT
COMMON /RDWEDG/ ANGLE(10), RADLE(5), WX1, DWX, WXI(20), TWEDGE, WK,
1 NWX, NANGLE, NRADLE, WEDGEM, AXISYM, ISW9B
COMMON /CHAN/ CPI(5), CP2(5), CP3(5), CP4(5), CP5(5)
COMMON /AVG/ WSAVE
COMMON /TNE/ TN1(186), TN2(186)
COMMON /RDMOD/ LEWIS, IAMBIP
COMMON /READAT/ PRESAI, DELT11, TPRNTI, DELTXI, CXMAXI,
1 CTAPI, XZEROT, PARAMI(3,12,2), BZEROI,
2 HSTAG, READXS, READG, AAMS, AXIMOD,
3 ICASE, NEELS, ISW7B, INT, ICHAN, NOSI, INEQVI, NRECO, JCS(10)

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ORIGINAL PAGE IS OF POOR QUALITY

DATA	SP8 /3HND+	3.1.1.1.1.5.1.6.1.-1.1.1.1.1.3.3.3.97385.3.749384E-4.	NO+
1	-6.062030E-8.	4.637506E-12.-1.107704E-16.4.200563.236660.	NO+
2	2.3841.3373.	4.1.1.1.1.6.3.2.6*0.0.106000.160000.	NO+
3	20000.6*0.3*0./		NO+
DATA	SP9 /2HN+	2.1.1.1.5.1.0.-1.1.0.6*0.447600.1.2943.	N+
1	0.7.1.0.1.3.5.1.5.1.5.3*0.3*0./		N+
2	93456.134860.	263740.3*0.3*0./	N+
DATA	SP10 /2HO+	2.1.1.1.6.1.0.-1.1.0.6*0.372940.1.4938.0.	O+
1	3.1.0.4.10.6.7*0.0.76670.115700.7*0.3*0./		O+
DATA	SP11 /3HN2+	2.1.1.1.5.1.0.-1.1.2.0.3.23806.4.47257E-4.	N2+
1	3.95890E-8.1.52963E-12.-0.21145E-16.4.95160.357680.		N2+
2	2.3763.3124.4.1.1.1.2.4.2.2.6*0.0.25890.72797.		N2+
3	184760.6*0.3*0./		N2+
DATA	SP12 /3HO2+	2.1.1.1.6.1.0.-1.1.2.0.3.49213.3.37873E-4.	O2+
1	5.20841E-8.4.16207E-12.-0.97275E-16.4.66750.288000.2.		O2+
2	0.317.2628.5.1.1.1.2.2.8.4.4.5*0.0.558.91206.		O2+
3	109760.128860.5*C.3*0./		O2+
DATA	SP13 /3HCC2.	2.1.4.1.6.1.0.1.2.0.6*0.0.0000.3.1.8958.	CO2
1	1977.1.1.0.1.9*0.10*0.960.960.3380./		CO2
DATA	SP14 /2HCO.	2.1.4.1.6.1.0.2*1.0.3.39468.3.22824E-4.	CO
1	3.94364E-8.2.17519E-12.-0.42966E-16.4.204.		CO
2	66770.2.3169.3083.5.1.1.1.6.3.6.2.5*0.0.		CO
3	139200.159830.178120.186055.5*0.3*0./		CO
DATA	SP15 /2HCN.	2.1.4.1.5.1.0.2*1.0.3.25545.4.33773E-4.	CN
1	3.93346E-8.1.59712E-12.-0.23789E-16.5.6334.		CN
2	197170.2.2226.2939.8.1.1.1.2.4.2.4.4.2*0.		CN
3	0.26069.7375.154263.168570.170640.174230.185220.		CN
4	2*0.3*0./		CN
DATA	SP16 /2HHE.	1.1.3.1.2*0.1.2*0.6*0.0.1.-1.5846.0.1.1.0.	HE
1	1.9*0.10*0.3*0./		HE
DATA	SP17 /1HC.	1.4.1.2*0.1.2*0.6*0.263550.1.0637.0.	C
1	10.1.0.9.5.1.5.9.3.15.34.99.401.25.29146.61894.		C
2	96452.172580.177213.183240.200100.214400.243000.3*0./		C
DATA	SP18 /2HC+	2.1.1.1.4.1.0.-1.1.0.6*0.523310.1.0636.0.	C+
1	4.1.0.2.4.12.10.6*0.0.183.123040.214240.6*0.3*0./		C+
DATA	SP19 /3HHE+	2.1.1.1.3.1.0.-1.1.0.6*0.566840.1.-1.5846.	HE+
1	0.1.1.0.2.9*0.10*0.3*0./		HE+
DATA	SP20 /3HAR+	2.1.1.1.7.1.0.-1.1.0.6*0.363330.1.	AR+
1	1.8663.0.2.1.0.4.2.8*0.0.4094.0*0.3*0./		AR+
DATA	SP21 /4HHE3S.	1.1.3.1.2*0.1.2*0.6*0.455910.1.-1.5846.0.	HE3S
1	1.1.0.3.9*0.10*0.3*0./		HE3S
DATA	SP22 /4HHE1S.	1.1.3.1.2*0.1.2*0.6*0.475260.1.-1.5846.0.	HE1S
1	1.1.0.1.9*0.10*0.3*0./		HE1S
DATA	SP23 /4HHE2+	2.1.1.1.3.1.0.-1.1.2.0.6*0.511720.2.	HE2+
1	3.5619.2343.1.1.0.2.9*0.10*0.3*0./		HE2+
DATA	SP24 /3HHE2.	1.1.3.1.2*0.2.2*0.6*0.413650.2.-3.6287.	HE2
1	2492.1.1.0.3.9*0.10*0.3*0./		HE2
DATA	SP25 /3HCO+	3.1.1.1.4.1.0.-1.1.2*1.3.49411.2.10083E-4.	CO+
1	1.11714E-8.0.58582E-12.0.13605E-16.4.2967.		CO+
2	389950.2.2931.3142.3.1.1.1.2.4.2.7*0.0.59278.		CO+
3	131166.7*0.2*0./		CO+
DATA	SP26 /4HAR*M.	1.1.7.1.2*0.1.2*0.6*0.266350.1.1.8663.0.	AR*M
1	1.1.0.6.9*0.10*0.3*0./		AR*M
DATA	SP27 /4HAR*R.	1.1.7.1.2*0.1.2*0.6*0.267970.1.1.8663.0.	AR*R
1	1.1.0.6.9*0.10*0.3*0./		AR*R

DATA	SP28 /4HAR2+	2.1.1.1.7.1.0.0.-1.2.0.0.6*0.337040.2.3.597.	AR2+
1	AIR-1	115.1.1.0.2.9*0.10*0.3*0./	AR2+
DATA	GPI /5HAIR-1	3.1.11.1.26.1.5.1.1.1.5.1.6.1.7*0.78823.	AIR-1
1		21177.8*0.1.1.5.1.2.1.3.1.7.1.1.	AIR-1
2		8.1.9.1.10.1.11.1.12.1.9*0.1.1.2.1.3.1.4.1.5.1.6.1.7.1.8.1.	AIR-1
3		9.1.10.1.11.1.12.1.13.1.14.1.15.1.16.1.17.1.18.1.19.1.20.1.	AIR-1
4		21.1.22.1.23.1.24.1.25.1.26.1.27.1.28.1.29.1.30.1.31.1.32.1.	AIR-1
5		5.1.0.0.1.1./	AIR-1
C	AIR-2	IS LOW-TEMP. MODEL WITH NO+ THE SOLE ION SPECIES.	AIR-2
DATA	GP2 /5HAIR-2	3.1.7.1.15.1.1.1.1.1.1.5.1.6.1.7*0.78823.	AIR-2
1		21177.8*0.1.1.5.1.2.1.3.1.7.1.1.	AIR-2
2		8.1.13*0.1.1.2.1.3.1.4.1.5.1.6.1.7.1.8.1.9.1.10.1.11.1.12.1.	AIR-2
3		19.1.20.1.21.22.1.23.1.24.1.25.1.26.1.27.1.28.1.29.1.30.1.31.1.32.1.	AIR-2
C	NONEQUILIBRIUM MODEL FOR ARGON		ARGON
DATA	GP3 /SHARGON	2.1.6.1.17.1.2.1.1.1.7.1.8*0.1.9*0.1.1.4.1.	ARGON
1		26.1.27.1.28.1.29.1.30.1.31.1.32.1.33.1.34.1.35.1.36.1.37.1.38.1.39.1.40.1.	ARGON
2		83.1.84.1.85.1.86.1.87.1.88.1.89.1.90.1.91.1.92.1.93.1.94.1.95.1.96.1.97.1.98.1.99.1.100.1.	ARGON
3		4.1.9*0.1.1.20.1.4.1.2.1.2.1./	ARGON
C	NONEQUILIBRIUM MODEL FOR HELIUM		HELIUM
DATA	GP4 /6HHELIUM	2.1.7.1.19.1.2.1.1.1.3.1.8*0.1.9*0.1.1.16.1.	HELIUM
1		21.1.22.1.23.1.24.1.25.1.26.1.27.1.28.1.29.1.30.1.31.1.32.1.33.1.34.1.35.1.36.1.37.1.38.1.39.1.40.1.	HELIUM
2		41.1.42.1.43.1.44.1.45.1.46.1.47.1.48.1.49.1.50.1.51.1.52.1.	HELIUM
3		53.1.45*0.16.1.9*0.1.1.19.1.16.1.1.2.1./	HELIUM
C	PLANETARY ATMOSPHERE MODEL - MOLE FRACS CO2. 0.2 AR. 0.05 N2		CONAR
DATA	GP5 /5HCONAR	5.1.19.1.48.1.8.1.1.1.4.1.5.1.6.1.7.1.5*0.	CONAR
1		0.75.0.2.0.05.7*0.1.1.4.1.13.1.5.1.6.1.2.1.3.1.7.1.14.1.	CONAR
2		15.1.17.1.8.1.9.1.10.1.11.1.12.1.18.1.20.1.25.1.0.	CONAR
3		1.1.2.1.3.1.4.1.5.1.6.1.7.1.8.1.9.1.10.1.11.1.12.1.13.1.14.1.	CONAR
4		15.1.16.1.17.1.18.1.19.1.20.1.21.1.22.1.23.1.24.1.25.1.26.1.	CONAR
5		54.1.55.1.56.1.57.1.58.1.59.1.60.1.61.1.62.1.63.1.64.1.65.1.	CONAR
6		66.1.67.1.68.1.69.1.70.1.71.1.72.1.73.1.74.1.75.1.16*0.	CONAR
7		13.1.4.1.5.1.7*0.3.1.14.1.0.2.1./	CONAR
C	PLANETARY ATMOSPHERE MODEL FOR TEMPERATURES TO 7000 DEG. K		CONAR2
DATA	GP6 /6HCONAR2	5.1.14.1.33.1.3.1.1.1.4.1.5.1.6.1.7.1.5*0.	CONAR2
1		75.2.05.7*0.1.1.4.1.13.1.5.1.6.1.2.1.3.1.7.1.14.1.15.1.	CONAR2
2		17.1.8.1.18.1.25.1.6*0.1.1.2.1.3.1.4.1.5.1.6.1.7.1.8.1.9.1.	CONAR2
3		10.1.11.1.12.1.19.1.20.1.26.1.54.1.55.1.56.1.57.1.58.1.59.1.	CONAR2
4		60.1.61.1.62.1.65.1.66.1.67.1.68.1.70.1.72.1.73.1.74.1.75.1.	CONAR2
5		31*0.13.1.4.1.5.1.7*0.3.1.14.1.0.2.1./	CONAR2
DATA	RP1 /3.6E14	-1.1.117980.1.1.1.1.1.6.1.2*0.3.1.2*0.1.2*0./	REAC 001
1		2.2*0.7.1.2.1.7.1.4.1.13.1.14.1.15.1.17.1.3*0./	REAC 001
DATA	RP2 /9.E15	-1.1.117980.0.0.2.1.1.1.6.1.3.1.0.3.1.2*0.1.1.1.	REAC 002
1		0.3.2*0.0.10*0./	REAC 002
DATA	RP3 /3.2E15	-1.1.117980.0.0.1.1.2.1.6.1.2*0.3.1.6.1.0.2.	REAC 003
1		2*0.2.1.0.0.10*0./	REAC 003
DATA	RP4 /7.2E14	-1.1.117980.0.0.2.1.2.1.6.1.5.1.0.3.1.5.1.0.	REAC 004
1		1.1.0.2.1.0.0.10*0./	REAC 004
DATA	RP5 /1.9E15	-0.5.225040.1.1.1.1.1.5.1.2*0.2.1.2*0.1.	REAC 005
1		2*0.2.2*0.8.1.3.1.7.1.6.1.4.1.13.1.14.1.15.1.17.1.2*0./	REAC 005
DATA	RP6 /4.1E16	-1.5.225040.0.0.2.1.1.1.5.1.2.1.0.2.1.2*0.1.	REAC 006
1		1.0.3.2*0.0.10*0./	REAC 006
DATA	RP7 /4.7E15	-0.5.225040.0.0.1.1.2.1.5.1.2*0.2.1.5.1.0.2.	REAC 007
1		2*0.2.1.0.0.10*0./	REAC 007
DATA	RP8 /3.9E14	-1.5.150050.1.1.1.1.2.1.7.1.2*0.2.1.3.1.0.1.	REAC 008

1 2*0.2*1.0.0.6.1.5.1.6.1.4.1.13.1.14.1.15.1.4*0./ REAC 008
 1 DATA RP9 /7.8E14,-1.5.150C50.1.1.1.2.1.7.1.2*0.2.1.3.1.0.1. REAC 009
 1 2*0.2*1.0.4.1.3.1.2.1.7.1.17.1.6*0./ REAC 010
 1 DATA RP10 /3.2E13,1.39150.0.2.1.2.1.3.1.7.1.0.2.1.6.1.0.1. REAC 011
 1 1.0.1.0.0.0.0.10*0./ REAC 012
 1 DATA RP11 /7.8E13,0.75510.0.2.1.2.1.3.1.5.1.0.2.1.7.1.0.1. REAC 013
 1 1.0.1.0.0.0.0.10*0./ REAC 014
 1 DATA RP12 /6.7E15,-1.5.0.0.2.1.2.1.8.1.1.1.0.2.1.3.1.0.1.1. REAC 015
 1 0.1.1.0.0.0.10*0./ REAC 016
 1 DATA RP13 /2.2E22,-4.5.0.0.2.1.2.1.10.1.1.1.0.3.1.1.1.0.1.1. REAC 017
 1 2.0.1.1.0.0.0.10*0./ REAC 018
 1 DATA RP14 /2.2E22,-4.5.0.0.2.1.2.1.9.1.1.1.0.2.1.1.1.0.1.1. REAC 019
 1 2.0.1.1.0.0.0.10*0./ REAC 020
 1 DATA RP15 /8.8E15,-1.5.0.0.2.1.1.1.12.1.1.1.0.3.1.2*0.1.1.0. REAC 021
 1 2.2*0.0.10*0./ REAC 022
 1 DATA RP16 /7.8E13,0.5.0.0.2.1.2.1.6.1.10.1.0.3.1.12.1.0.1.1. REAC 023
 1 1.0.1.1.0.0.0.10*0./ REAC 024
 1 DATA RP17 /7.8E13,0.5.0.0.2.1.2.1.2.1.11.1.0.5.1.9.1.0.1.1. REAC 025
 1 0.1.1.0.0.0.10*0./ REAC 026
 1 DATA RP18 /1.5E16,-1.5.0.0.2.1.1.1.1.1.1.1.0.2.1.2*0.1.1.1. REAC 027
 1 0.2.2*0.0.10*0./ REAC 028
 1 DATA RP19 /1.8E14,-2.5.0.0.3.1.2.1.7.1.8.1.1.1.5.1.6.1.0.1.1. REAC 029
 1 1.1.1.0.0.0.10*0./ REAC 030
 1 DATA RP20 /2.2E16,-2.5.0.0.3.1.2.1.8.1.1.1.5.1.7.1.5.1.0.1.1. REAC 031
 1 1.1.1.0.0.0.10*0./ REAC 032
 1 DATA RP21 /1.5E13,0.0.0.2.1.2.1.7.1.10.1.0.3.1.8.1.0.1.1.1. REAC 033
 1 0.1.1.0.0.0.10*0./ REAC 034
 1 DATA RP22 /3.4E11,-2.45700.0.2.1.2.1.5.1.10.1.0.3.1.11.1.0. REAC 035
 1 1.1.0.0.1.0.0.10*0./ REAC 036
 1 DATA RP23 /4.8E14,0.0.0.2.1.2.1.1.9.1.0.2.1.8.1.0.1.1.0. REAC 037
 1 1.1.0.0.10*0./ REAC 038
 1 DATA RP24 /1.8E15,0.5.0.0.2.1.2.1.7.1.12.1.0.6.1.8.1.0.1.1. REAC 039
 1 0.1.1.0.0.0.10*0./ REAC 040
 1 DATA RP25 /1.8E14,0.0.0.2.1.2.1.6.1.9.1.0.3.1.8.1.0.1.1.0. REAC 041
 1 1.1.0.0.0.10*0./ REAC 042
 1 DATA RP26 /8.8E16,-2.5.0.0.3.1.2.1.8.1.1.1.6.1.7.1.6.1.0.1.1. REAC 043
 1 1.1.1.0.0.0.10*0./ REAC 044
 1 DATA RP35 /5.46E21,-4.3.0.0.1.2.1.1.1.1.1.19.1.0.21.1.2*0.2*1. REAC 045
 1 0.1.2*0.1.1.1.9*0./ REAC 046
 1 DATA RP36 /1.82E21,-4.3.0.0.1.2.1.1.1.1.1.19.1.0.22.1.2*0.2*1. REAC 047
 1 0.1.2*0.1.1.1.9*0./ REAC 048
 1 DATA RP37 /1.27E11,-0.81.0.0.2.1.1.1.1.1.19.1.0.21.1.2*0.1. REAC 049
 1 1.0.1.2*0.0.10*0./ REAC 050
 1 DATA RP38 /3.80E10,-0.85.0.0.2.1.1.1.1.1.19.1.0.22.1.2*0.2*1. REAC 051
 1 0.1.2*0.0.10*0./ REAC 052
 1 DATA RP39 /3.92E16,2*0.1.2.1.1.1.16.1.19.1.0.23.1.2*0.2*1. REAC 053
 1 0.1.2*0.1.1.16.1.9*0./ REAC 054
 1 DATA RP40 /1.54E21,-4.3.0.0.1.2.1.1.1.1.1.23.1.0.24.1.2*0.1.1. REAC 055
 1 0.1.2*0.1.1.1.9*0./ REAC 056
 1 DATA RP41 /5.13E20,-4.3.0.0.1.2.1.1.1.1.1.23.1.0.16.1.2*0.2*1. REAC 057
 1 2*0.2*0.1.1.1.9*0./ REAC 058
 1 DATA RP42 /2.26E14,3*0.2*2.1.1.1.23.1.0.16.1.21.1.0.2*1.0. REAC 059
 1 1.1.0.0.0.10*0./ REAC 060
 1 DATA RP43 /7.5E13,3*0.2*2.1.1.1.23.1.0.16.1.22.1.0.2*1.0.1. REAC 061
 1 1.0.0.0.10*0./ REAC 062

DATA	RP44	/3.65E16,-0.5*0.,1.,2*1.1,22.1,2*0.,21.1,2*0.,1.,2*0.,	REAC	044
1	1.	2*0.,1.,1.1,1.9*0./	REAC	044
1	DATA	RP45 /8.E14,-0.25,1272.,1.,2*1.1,21.1,2*0.,16.1,2*0.,1.,2*0.,	REAC	045
1	1.	2*0.,1.,1.1,1.9*0./	REAC	045
1	DATA	RP46 /8.E14,-0.25,1272.,1.,2*1.1,22.1,2*0.,16.1,2*0.,1.,2*0.,	REAC	046
1	1.	2*0.,1.,1.1,1.9*0./	REAC	046
1	DATA	RP47 /5.20E10,0.,1590.,1.,2*1.1,22.1,2*0.,16.1,2*0.,1.,2*0.,	REAC	047
1	1.	2*0.,1.,1.1,1.9*0./	REAC	047
1	DATA	RP48 /1.87E15,0.,167.0,0.,1.1,3.1,21.1,2*0.,1.1,16.1,19.1,	REAC	048
1	2*0.,3*1.0,10*0./	REAC	048	
1	DATA	RP49 /5.05E15,0.,167.2*0.,2.1,3.1,21.1,22.1,0.,1.1,16.1,	REAC	049
1	19.1,2*1.0,3*1.0,10*0./	REAC	049	
1	DATA	RP50 /5.28E15,0.,167.2*0.,1.1,3.1,22.1,2*0.,1.1,16.1,19.1,	REAC	050
1	2*0.,3*1.0,10*0./	REAC	050	
1	DATA	RP51 /5.2E14,0.,5.0,1.,2.1,1.1,16.1,21.1,0.,24.1,2*0.,2*1.,	REAC	051
1	0.,1.,2*0.,1.1,1.16,1.9*0./	REAC	051	
1	DATA	RP52 /1.87E15,0.,167.2*0.,1.1,3.1,24.1,2*0.,1.1,16.1,19.1,	REAC	052
1	2*0.,1.,3.1,1.0,10*0./	REAC	052	
1	DATA	RP53 /8.E14,-0.25,1272.,1.,2*1.1,24.1,2*0.,16.1,2*0.,1.,2*0.,	REAC	053
1	2*0.,1.,1.1,1.9*0./	REAC	053	
1	DATA	RP54 /4.48E15,-1.,256000.,1.,1.1,2.1,14.1,2*0.,17.1,3.1,0.,	REAC	054
1	1.,2*0.,2*1.0,9.1,2.1,3.1,4.1,5.1,6.1,13.1,14.1,15.1,17.1,	REAC	054	
2	C./	REAC	054	
1	DATA	RP55 /8.81E14,-2.,125600.,1.,1.1,2.1,13.1,2*0.,3.1,14.1,0.,	REAC	055
1	1.,2*0.,2*1.0,9.1,2.1,3.1,4.1,5.1,6.1,13.1,14.1,15.1,17.1,	REAC	055	
2	0./	REAC	055	
1	DATA	RP56 /2.33E11,5.,130500.,0.,1.1,2.1,14.1,2*0.,13.1,17.1,0.,	REAC	056
2*0.,2*1.0,11*0./	REAC	056		
1	DATA	RP57 /2.73E13,5.,138100.,0.,2.1,2.1,3.1,14.1,0.,6.1,17.1,0.,	REAC	057
2*1.0,2*1.0,11*0./	REAC	057		
1	DATA	RP58 /2.86E13,5.,106500.,0.,2.1,2.1,2.1,14.1,0.,7.1,17.1,0.,	REAC	058
2*1.0,2*1.0,11*0./	REAC	058		
1	DATA	RP59 /4.59E10,5.,23970.,0.,2.1,2.1,7.1,14.1,0.,13.,2.1,0.,	REAC	059
2*1.0,2*1.0,11*0./	REAC	059		
1	DATA	RP60 /2.55E11,5.,7606.,0.,2.1,2.1,13.1,3.1,0.,6.1,14.1,0.,	REAC	060
2*1.0,2*1.0,11*0./	REAC	060		
1	DATA	RP61 /1.07E14,5.,67050.,0.,2.1,2.1,14.1,25.1,0.,13.1,18.1,0.,	REAC	061
2*1.0,2*1.0,11*0./	REAC	061		
1	DATA	RP62 /6.03E13,5.,63360.,0.,2.1,2.1,14.1,18.1,0.,17.1,25.1,0.,	REAC	062
2*1.0,2*1.0,11*0./	REAC	062		
1	DATA	RP63 /6.66E14,5.,54160.,0.,2.1,2.1,3.1,18.1,0.,17.1,10.1,0.,	REAC	063
2*1.0,2*1.0,11*0./	REAC	063		
1	DATA	RP64 /1.09E14,5.,9222.,0.,2.1,2.1,14.1,10.1,0.,3.1,25.1,0.,	REAC	064
2*1.0,2*1.0,11*0./	REAC	064		
1	DATA	RP65 /5.47E14,5.,74700.,0.,2.1,2.1,3.1,25.1,0.,6.1,18.1,0.,	REAC	065
2*1.0,2*1.0,11*0./	REAC	065		
1	DATA	RP66 /1.5E16,-1.5,0.,0.,2.1,2.1,25.1,1.1,0.,17.1,3.1,0.,	REAC	065
2*1.0,2*1.0,11*0./	REAC	065		
1	DATA	RP67 /2.2E22,-4.5,0.,0.,2.1,2.1,18.1,1.1,0.,17.1,1.1,0.,	REAC	067
1.,2.,0.,2.22.,11*0./	REAC	067		
1	DATA	RP68 /4.59E14,5.,109,00.,0.,2.1,2.1,14.1,8.1,0.,25.1,7.1,0.,	REAC	068
2*1.0,2*1.0,11*0./	REAC	068		
1	DATA	RP69 /4.53E14,5.,44490.,0.,2.1,2.1,14.1,12.1,0.,25.1,6.1,0.,	REAC	069
2*1.0,2*1.0,11*0./	REAC	069		
1	DATA	RP70 /5.96E14,5.,3227.,0.,2.1,2.1,17.1,8.1,0.,25.1,2.1,0.,	REAC	070

1	DATA	2*1.0.0.2*1.0.0.11*0.0.	2.1.2.1.2.0.1.1.0.0.4.1.1.1.0.0.	REAC	070
1	RP71	/2.2E22,-4.5.0.0.11*0.0.	2.1.2.1.2.0.1.1.0.0.4.1.1.1.0.0.	REAC	071
1	DATA	1.0.0.2*1.0.0.11*0.0.	1.1.2.1.1.15.1.2*0.0.17.1.2.1.0.0.	REAC	072
1	RP72	/3.6E15,-1.131160.1.1.1.2.1.4.1.5.1.6.1.7.1.13.1.14.1.15.1.	1.1.2.1.1.15.1.2*0.0.17.1.2.1.0.0.	REAC	073
2	DATA	17.1/4.3E14,.5.69670.0.0.2.1.2.1.1.4.1.2.1.0.0.15.1.3.1.0.0.	1.1.2.1.1.15.1.2*0.0.17.1.2.1.0.0.	REAC	074
1	RP73	/4.3E14,.5.69670.0.0.11*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	075
1	DATA	2*1.0.0.2*1.0.0.11*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	076
1	RP74	/1.5E13,.5.51670.0.0.2.1.2.1.5.1.17.1.0.0.15.1.2.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	077
1	DATA	2*1.0.0.2*1.0.0.11*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	078
1	RP75	/2.E14,.1.55640.0.0.2.1.2.1.1.15.1.3.1.0.0.17.1.7.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	079
1	DATA	2*1.0.0.2*1.0.0.11*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	080
1	RP76	/3.64E21,-4.3.0.0.0.2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	081
1	DATA	2.1.0.0.1.1.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	082
1	RP77	/3.64E21,-4.3.0.0.0.2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	083
1	DATA	2.1.0.0.1.1.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	084
1	RP78	/8.22E10,-0.81.0.0.0.2.1.1.1.1.1.20.1.0.0.26.1.2*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	085
1	DATA	1.1.0.0.1.1.2*0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	086
1	RP79	/8.22E10,-0.81.0.0.0.2.1.1.1.1.1.20.1.0.0.27.1.2*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	087
1	DATA	1.1.0.0.1.1.2*0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	088
1	RP80	/6.E10,-0.5.0.0.0.2.1.1.1.1.1.20.1.0.0.4.1.2*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	089
1	DATA	1.1.0.0.1.1.2*0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	090
1	RP81	/5.E14,.0.5.0.0.0.2.1.2.1.1.1.26.1.0.0.1.1.4.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	091
1	DATA	1.1.0.0.1.1.1.0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	092
1	RP82	/7.2E13,.0.5.0.0.0.2.1.2.1.1.1.27.1.0.0.1.1.4.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	093
1	DATA	1.1.0.0.1.1.1.0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	094
1	RP83	/8.E4,.0.0.0.0.0.2.1.1.1.1.1.27.1.2*0.0.4.1.2*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	095
1	DATA	1.2*0.0.1.1.2*0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	096
1	RP84	/1.E17,-0.5.0.0.0.0.2.1.2.1.1.1.27.1.0.0.1.1.26.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	097
1	DATA	1.1.0.0.1.1.1.0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	098
1	RP85	/3.5E9,.0.5.0.0.0.2.1.1.1.4.1.26.1.0.0.4.1.2*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	099
1	DATA	1.1.0.0.2.1.2*0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	100
1	RP86	/8.7E14,-0.56.0.0.0.2.1.1.1.1.4.1.26.1.0.0.4.1.2*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	101
1	DATA	2.1.0.0.3.2*0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	102
1	RP87	/3.5E9,.0.5.0.0.0.2.1.1.1.4.1.27.1.0.0.4.1.2*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	103
1	DATA	1.1.0.0.2.1.2*0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	104
1	RP88	/8.7E14,-0.56.0.0.0.2.1.1.1.1.4.1.27.1.0.0.4.1.2*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	105
1	DATA	1.1.0.0.3.2*0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	106
1	RP89	/5.2E15,-0.75.0.0.0.2.1.2.1.1.4.1.20.1.0.0.4.1.28.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	107
1	DATA	2.1.0.0.1.1.0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	108
1	RP90	/2.8E16,-0.67.1252.0.0.2.1.2.1.1.1.28.1.0.0.4.1.26.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	109
1	DATA	1.1.0.0.1.1.0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	110
1	RP91	/2.8E16,-0.67.1252.0.0.2.1.2.1.1.1.28.1.0.0.4.1.27.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	111
1	DATA	1.1.0.0.1.1.0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	112
1	RP92	/2.E21,-4.3.0.0.0.2.1.2.1.1.1.28.1.0.0.1.1.4.1.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	113
1	DATA	2.1.0.0.1.1.2.0.0.0.10*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	114
1	RP93	/4*2.1.1.1.1.7*2.1.1.6*1.1.2.1.6*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	115
1	DATA	2*2.1.2*0.0.1.1.2*2.1.2*1.1.3*2.1.0.0.3*2.1.1.1.2*2.1.6*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	116
2	DATA	2*5.1.2*0.0.1.3.1.2*5.1.2*4.1.3*5.1.6.1.3*5.1.2*5.1.6*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	117
3	DATA	109890.0.91540.0.109890.0.91540.0.0.98040.0.458270.0.2*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	118
4	DATA	18350.0.456730.0.2*475080.0.346840.0.365190.0.383540.0.0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	119
5	DATA	260350.0.413480.0.6*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	120
6	DATA	25*0.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	121
7	DATA	0.0.100.0.200.0.400.0.600.0.800.0.1000.0.1500.0.2000.0.3000.0.4000.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	122
8	DATA	5000.0.6000.0.7000.0.8000.0.9000.0.10000.0.12000.0.14000.0.16000.0.	2.1.2.1.1.1.20.1.0.0.1.1.26.1.0.0.	REAC	123

9	18000	20000	25000	30000	35000	40000	45000	50000		TNE/HE
A	70000	100000								TNE/HE
B	0.5E-16	5.59E-16	5.83E-16	5.99E-16	6.11E-16	6.21E-16	6.34E-16	6.47E-16		TNE/HE
C	6.39E-16	6.51E-16	6.67E-16	6.76E-16	6.81E-16	6.84E-16	6.87E-16	6.90E-16		TNE/HE
D	6.84E-16	6.83E-16	6.80E-16	6.77E-16	6.74E-16	6.71E-16	6.68E-16	6.65E-16		TNE/HE
E	6.42E-16	6.29E-16	6.15E-16	6.00E-16	5.80E-16	5.46E-16	5.14E-16	4.82E-16		TNE/HE
F	4.85E-16	4.60E-16	4.32E-16	4.04E-16	3.74E-16	3.42E-16	3.08E-16	2.74E-16		TNE/HE
G	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4		TNE/HE
	DATA	1.4	1.5	1.5	1.5	1.5	1.5	1.5		TNE/AR
1	2*2.1	3*0.2	2*1.0	2*1.0	1.0	1.0	1.0	1.0		TNE/AR
2	2*5.1	3*1.2	5*1.6	1.3	1.6	1.3	1.3	1.3		TNE/AR
3	96970	95360	96970	95360	363330	266350	267970	267970		TNE/AR
4	16000	0	226000	0	706800	690700	337040	8*0		TNE/AR
5	2*0.2	0.7	1.20	0						TNE/AR
6	0.25	0.50	1.00	2.00	3.00	4.00	6.00	9.00		TNE/AR
7	400	1500	2000	2500	4000	6000	8000	10000		TNE/AR
8	15000	20000	25000	30000	35000	40000	45000	50000		TNE/AR
9	10000									TNE/AR
A	10E-16	10E-16	6E-16	3.57E-16	2.12E-16	1.39E-16	0.97E-16	0.7E-16		TNE/AR
B	0.57E-16	0.40E-16	0.35E-16	0.32E-16	0.29E-16	0.29E-16	0.29E-16	0.29E-16		TNE/AR
C	0.31E-16	0.38E-16	0.51E-16	0.98E-16	1.61E-16	2.40E-16	3.40E-16	4.40E-16		TNE/AR
D	3.08E-16	3.73E-16	4.65E-16	6.02E-16	7.19E-16	8.11E-16	9.11E-16	10.11E-16		TNE/AR
E	9.81E-16	9.30E-16	9.63E-16	9.82E-16	10.10E-16	10.10E-16	10.10E-16	10.10E-16		TNE/AR
	2E77									NOZZ
	DCA 0.75-INCH	NOZZLF								NOZZ
1	1.1	2.1	1.1	3.1	1.1	2.1	1.1	2.1		NOZZ
2	-5.568490	-5.455796	-1.523172	-1.264826	-1.099852	-1.099852	-1.099852	-1.099852		NOZZ
3	0.329700	0.0	0.0	0.0	0.0	0.0	0.0	0.0		NOZZ
4	-12.978253	-2.747477	0.0	2.364483	-5.449149	0.127000	0.127000	0.127000		NOZZ
5	1.951731	-0.052408	0.0	1.714493	-1.539789	0.317500	0.317500	0.317500		NOZZ
6	-0.317500	-1.732051	0.0	2.224500	0.0	1.270000	1.270000	1.270000		NOZZ
7	2.222500	0.0	1.270000	0.907699	0.267949	0.0	0.0	0.0		NOZZ
8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		NOZZ
9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		NOZZ
	4HDCA /									NOZZ
	DCA 1.5-INCH	NOZZLE								NOZZ
1	1.1	2.1	1.1	2.1	1.1	2.1	1.1	2.1		NOZZ
2	-9.451499	-8.304556	-0.006547	0.0	0.0	0.0	0.0	0.0		NOZZ
3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		NOZZ
4	-22.738861	-2.747476	0.0	3.609504	-8.238090	1.270000	1.270000	1.270000		NOZZ
5	1.904823	-0.052408	0.0	2.031999	0.0	0.0	0.0	0.0		NOZZ
6	2.031999	0.0	1.270000	1.900517	0.267949	0.0	0.0	0.0		NOZZ
7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		NOZZ
8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		NOZZ
9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		NOZZ
	4HDCA /									NOZZ
	MRA 2.25-INCH	NOZZLE								NOZZ
1	1.1	3.1	1.1	2.1	1.1	2.1	1.1	2.1		NOZZ
2	-2.658344	-2.447308	-1.127258	-0.610565	-0.006647	0.0	0.0	0.0		NOZZ
3	0.032870	0.0	0.0	0.0	0.0	0.0	0.0	0.0		NOZZ
4	5.171379	-0.176327	0.0	5.339946	-2.711272	0.304800	0.304800	0.304800		NOZZ
5	1.253495	-1.732051	0.0	3.524453	-0.577332	0.635000	0.635000	0.635000		NOZZ

6	2.857325	-0.052408	0.0	0.0	2.984499	C.0	0.127000	NOZZ	3
7	2.984499	0.0	0.127000	0.0	2.853019	0.0	0.0	NOZZ	3
8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	3
9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	3
A	MRA 1.00-INCH NOZZLE								
1	DATA ZP 4 /	1.27000	6.1	2.1	2.1	0.1	0.1	NOZZ	4
2	1.1	3.1	1.1	2.1	2.1	1.1	0.1	NOZZ	4
3	-3.529553	-3.308723	-1.075787	-0.559095	-0.006647	0.0	0.0	NOZZ	4
4	0.032870	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	4
5	0.015062	-0.176327	0.0	5.324564	-3.583686	0.0	0.317500	NOZZ	4
4	-0.247563	-1.732051	0.0	1.933256	-0.525861	0.0	0.635000	NOZZ	4
6	1.269825	-0.052408	0.0	1.397070	C.0	0.0	C.127000	NOZZ	4
7	1.297000	0.0	0.127000	1.265519	0.267949	0.0	0.0	NOZZ	4
8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	4
9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	4
A	MRA 1.00-INCH NOZZLE								
1	ED: 0.32-INCH NOZZLE								
2	DATA ZP 5 /	0.40640	6.1	4.1	4.1	3.1	0.1	NOZZ	5
3	1.1	3.1	1.1	2.1	2.1	1.1	0.1	NOZZ	5
4	-4.836277	-4.806674	-1.720592	-1.718576	-0.053673	0.0	0.0	NOZZ	5
5	0.058642	1.134806	1.014119	0.0	0.0	0.0	0.127000	NOZZ	5
6	1.725971	-0.176327	0.0	2.453567	-4.858330	0.0	0.127000	NOZZ	5
7	0.429518	-0.445229	0.0	1.079555	-1.772248	0.0	0.127000	NOZZ	5
8	0.393271	-0.466308	0.0	0.533400	0.0	0.0	C.127000	NOZZ	5
9	0.533400	0.0	0.127000	0.390222	0.520567	0.0	0.0	NOZZ	5
A	MRA 0.75-INCH NOZZLE								
1	ED: 0.75-INCH NOZZLE								
2	DATA ZP 6 /	0.98171	8.1	2.1	2.1	2.1	0.1	NOZZ	6
3	1.1	3.1	1.1	2.1	2.1	1.1	0.1	NOZZ	6
4	-5.974197	-5.544594	-2.863580	-2.851736	-2.792465	-2.735612	0.0	NOZZ	6
5	0.006647	0.0	0.052330	0.0	0.0	0.0	0.127000	NOZZ	6
6	1.509393	-0.176327	0.0	2.437734	-5.996250	0.0	0.127000	NOZZ	6
7	0.470487	-0.445229	0.0	1.065978	-2.915236	0.0	0.127000	NOZZ	6
8	0.981535	-0.577350	0.0	1.251729	-2.728965	0.0	0.127000	NOZZ	6
9	1.108710	-0.052408	0.0	1.108710	C.0	0.0	0.127000	NOZZ	6
A	MRA 0.75-INCH NOZZLE								
1	ED: 0.75-INCH NOZZLE								
2	DATA ZP 7 /	0.95250	6.1	2.1	2.1	2.1	0.1	NOZZ	7
3	1.1	3.1	1.1	2.1	2.1	1.1	0.1	NOZZ	7
4	-3.624795	-3.413759	-0.999293	-0.482600	-0.006647	0.0	0.0	NOZZ	7
5	0.032870	0.0	0.0	0.0	0.0	0.0	0.1	NOZZ	7
6	4.984842	-0.176327	0.0	5.323822	-3.677723	0.0	0.304800	NOZZ	7
7	-0.436579	-1.732050	0.0	1.511746	-0.449367	0.0	0.635000	NOZZ	7
8	0.952325	-0.052408	0.0	1.079499	0.0	0.0	C.127000	NOZZ	7
9	1.079499	0.0	0.127000	0.949019	0.267949	0.0	0.0	NOZZ	7
A	MRA 1.00-INCH NOZZLE								
1	ED: 1.00-INCH NOZZLE								
2	DATA ZP 8 /	1.90500	6.1	2.1	2.1	2.1	0.0	NOZZ	8
3	1.1	3.1	1.1	2.1	2.1	1.1	0.0	NOZZ	8
4	-3.17500	-3.17500	-3.17500	-3.17500	-3.17500	-3.17500	0.0	NOZZ	8

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1	1.1.	3.1.	1.1.	2.1.	1.1.	2.1.	1.1.	2.1.	1.1.	0.1.	0.1.	0.1.	NOZZ	8
2	-3.157434.	-2.946399.	0.0	-1.075492.	-0.558800.	-0.006647.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	8
3	0.032870.	0.0	0.0	0.0	5.338842.	-3.210362.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	8
4	5.082271.	-0.176327.	0.0	0.0	2.568238.	-0.525567.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	8
5	0.387933.	-1.732050.	0.0	0.0	2.031999.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	8
6	1.904823.	-0.052408.	0.0	0.0	1.900517.	0.267949.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	8
7	2.0031999.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	8
8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	8
9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	8
A	4HMRA /												NOZZ	8
10MW 2.25-INCH NOZZLE														
1	DATA ZP 9 /	2.85750.	3.1.	1.1.	2.1.	1.1.	2.1.	1.1.	2.1.	0.1.	0.1.	0.1.	NOZZ	9
2	-7.918200.	-7.698371.	0.0	-8.00100.	-5.714999.	-0.013283.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	9
3	0.039444.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	9
4	3.654496.	-0.176327.	0.0	0.0	4.738010.	-7.973334.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	9
5	-8.437203.	-1.732050.	0.0	0.0	5.570952.	-5.521296.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	9
6	2.856915.	-0.087489.	0.0	0.0	3.009897.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	9
7	3.009897.	0.0	0.0	0.152400.	2.852120.	0.267949.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	9
8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	9
9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	9
A	4H10MW /												NOZZ	9
EOS 1.09-INCH NOZZLE														
1	DATA ZP10 /	1.38176.	3.1.	1.1.	2.1.	1.1.	2.1.	1.1.	2.1.	0.1.	0.1.	0.1.	NOZZ	10
2	-3.540351.	-3.473112.	0.0	-3.55600.	-2.393882.	-1.280036.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	10
3	-0.005397.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	10
4	1.921388.	-0.176327.	0.0	0.0	2.295506.	-3.594458.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	10
5	0.829850.	-0.487733.	0.0	0.0	2.243805.	-2.319619.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	10
6	1.269022.	-0.305731.	0.0	0.0	1.903271.	-1.205774.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	10
7	1.381163.	-0.217432.	0.0	0.0	1.407157.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	10
8	1.407157.	0.0	0.0	0.0	1.380859.	0.267949.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	10
9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	10
A	4HEOS /												NOZZ	10
DCA T12 CHANNEL, PROFILE 1														
1	DATA ZP11 /	2.54000.	2.1.	1.1.	2.1.	1.1.	2.1.	1.1.	2.1.	0.1.	0.1.	0.1.	NOZZ	11
2	-6.730998.	-6.095999.	0.0	-7.62000.	-2.393882.	-1.280036.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	11
3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	11
4	-1.175995.	-0.577350.	0.0	0.0	3.809999.	-6.095999.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	11
5	2.539999.	0.0	0.0	0.0	105.079999.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	11
6	105.079999.	0.0	0.0	102.540000.	2.539999.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	11
7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	11
8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	11
9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	11
A	4H DCA /												NOZZ	11
DCA T12 CHANNEL, PROFILE 2														
1	DATA ZP12 /	1.27000.	2.1.	1.1.	2.1.	1.1.	2.1.	1.1.	2.1.	0.1.	0.1.	0.1.	NOZZ	12
2	-6.771664.	0.0	0.0	-7.62000.	-2.393882.	-1.280036.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	12
3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	12
4	-0.501576.	-0.495987.	0.0	0.0	16.509995.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	12
5	16.509995.	0.0	0.0	0.0	1.034893.	0.176327.	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	12
6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	12
7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	NOZZ	12

DATA VV	8/142..	1.1408.	1.0.	0..	0..	0..
DATA JJ	9/17.	0.	0.	0.	0.	0.
DATA VV	9/10.	51..	2..	0..	0..	0..
DATA II	10/2.	0.	0.	0.	0.	0.
DATA VV	10/998..	4.824.	0..	0..	0..	0..
DATA II	11/3.	0.	0.	0.	0.	0.
DATA JJ	11/10.	0.	0.	0.	0.	0.
DATA VV	11/997..	4.722.	0..	0..	0..	0..
DATA II	12/2.	0.	0.	0.	0.	0.
DATA JJ	12/2.	0.	0.	0.	0.	0.
DATA VV	12/1.0.	81..	20..	0..	0..	0..
DATA II	13/2.	2.	0.	0.	0.	0.
DATA JJ	13/15.	7.	5.	0.	0.	0.
DATA VV	13/1.0.	101..	20..	0..	0..	0..
DATA II	14/3.	0.	0.	0.	0.	0.
DATA JJ	14/3.	0.	0.	0.	0.	0.
DATA VV	14/1.0.	121..	20..	0..	0..	0..
DATA II	15/14.	13.	5.	6.	7.	7.
DATA JJ	15/14.	14.	14.	14.	14.	14.
DATA VV	15/1.0.	141..	20..	0..	0..	0..
DATA II	16/14.	13.	7.	13.	5.	5.
DATA JJ	16/15.	15.	13.	13.	13.	13.
DATA VV	16/1.0.	141..	20..	0..	0..	0..
DATA II	17/6.	7.	5.	5.	5.	5.
DATA JJ	17/13.	7.	7.	15.	5.	5.
DATA VV	17/1.0.	141..	20..	0..	0..	0..
DATA II	18/3.	3.	3.	2.	0.	0.
DATA JJ	18/5.	15.	7.	6.	0.	0.
DATA VV	18/1.0.	161..	20..	0..	0..	0..
DATA II	19/6.	6.	6.	5.	7.	7.
DATA JJ	19/15.	7.	15.	7.	7.	7.
DATA VV	19/1.0.	181..	20..	0..	0..	0..
DATA II	20/7.	5.	0.	0.	0.	0.
DATA JJ	20/7.	6.	0.	0.	0.	0.
DATA VV	20/1.0.	181..	20..	0..	0..	0..
DATA II	21/3.	0.	0.	0.	0.	0.
DATA JJ	21/3.	0.	0.	0.	0.	0.
DATA VV	21/1.0.	201..	20..	0..	0..	0..
DATA II	22/3.	3.	2.	3.	0.	0.
DATA JJ	22/15.	7.	6.	6.	0.	0.
DATA VV	22/1.0.	221..	20..	0..	0..	0..
DATA II	23/6.	6.	6.	7.	0.	0.
DATA JJ	23/15.	7.	6.	7.	0.	0.
DATA VV	23/1.0.	241..	20..	0..	0..	0..
DATA II	24/2.	2.	0.	0.	0.	0.
DATA JJ	24/15.	7.	0.	0.	0.	0.
DATA VV	24/2.0.	6.0.	0.5.	1.0.	1.0.	1.0.
DATA II	25/15.	7.	0.	0.	0.	0.
DATA JJ	25/0.5.	1.	1.	0.	0.	0.
DATA VV	25/2.	2.	0.	0.	0.	0.
DATA II	26/5.	3.	0.	0.	0.	0.
DATA JJ	26/1.0.	1.	1.	0.	0.	0.

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BD2 163
BD2 164
BD2 165

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42.8000E	04.3.0000E	04.1.0000E	03.2.0000E	03.3.0000E	03.4.0000E	03.802
55.0000E	03.5.0000E	03.7.0000E	03.8.0000E	03.9.0000E	03.1.0000E	04.802
61.2000E	04.1.4000E	04.1.6000E	04.1.8000E	04.2.0000E	04.2.2000E	04.802
72.4000E	03.4.2000E	03.5.0000E	03.6.0000E	03.7.0000E	03.8.0000E	03.802
83.0000E	03.1.0000E	03.2.0000E	03.3.0000E	03.4.0000E	03.5.0000E	04.802
99.0000E	04.2.2000E	04.2.4000E	04.2.6000E	04.2.8000E	04.3.0000E	04.802
\$1.0000E	03.8.0000E	03.9.0000E	03.1.0000E	03.2.0000E	03.3.0000E	03.802
\$1.6000E	04.1.8000E	04.3.0000E	04.2.0000E	04.1.2000E	04.1.4000E	04.802
\$2.0000E	04.3.0000E	04.1.0000E	03.2.0000E	03.3.0000E	03.4.0000E	03.802
\$2.0000E	04.1.0000E	04.1.6000E	03.8.0000E	03.9.0000E	03.1.0000E	04.802
\$2.4000E	04.2.6000E	04.2.8000E	04.3.0000E	04.2.0000E	04.2.2000E	04.802
DATA TL5/						
1 70*0.0	03.4.0000E	03.5.0000E	03.6.0000E	03.7.0000E	03.8.0000E	03.802
39.0000E	03.1.0000E	04.1.2000E	04.1.4000E	04.1.6000E	04.1.8000E	03.802
42.0000E	04.2.2000E	04.2.4000E	04.2.6000E	04.2.8000E	04.3.0000E	04.802
5 10*0.0						
DATA TL6/						
11.0000E	01.2.0000E	01.3.0000E	01.4.0000E	01.5.0000E	01.6.0000E	01.802
27.0000E	01.8.0000E	01.9.0000E	01.1.0000E	01.2.0000E	01.3.0000E	01.802
31.6000E	00.1.8000E	00.2.0000E	00.2.5000E	00.3.0000E	00.3.5000E	00.802
44.0000E	00.5.0000E	00.6.0000E	00.7.0000E	00.8.0000E	00.9.0000E	00.802
51.0000E	01.1.2000E	01.1.4000E	01.1.6000E	01.1.8000E	01.2.0000E	01.802
62.5000E	01.3.0000E	01.3.5000E	01.4.0000E	01.5.0000E	01.7.5000E	01.802
71.0000E	02.63*0.0					01.802
DATA TL11/						
11.8756E	01.1.9016E	01.1.9259E	01.1.9484E	01.1.9693E	01.1.9888E	01.802
22.0070E	01.2.0239E	01.2.0397E	01.2.0545E	01.2.0684E	01.2.0815E	01.802
32.0937E	01.2.1053E	01.2.1162E	01.2.1265E	01.2.1362E	01.2.1455E	01.802
42.1543E	01.2.1626E	01.2.1705E	01.2.1781E	01.2.1853E	01.2.1988E	01.802
52.2111E	01.2.2224E	01.2.2329E	01.2.2425E	01.2.2515E	01.2.2598E	01.802
62.2676E	01.2.2749E	01.2.2817E	01.2.2881E	01.2.2941E	01.2.2998E	01.802
72.3052E	01.2.3151E	01.2.3240E	01.2.3320E	01.2.3394E	01.2.3461E	01.802
82.3522E	01.2.3579E	01.2.3631E	01.2.3680E	01.2.3725E	01.2.3767E	01.802
92.3806E	01.2.3825E	01.2.3850E	01.2.3875E	01.2.3900E	01.2.3925E	01.802
\$1.2240E	01.1.1570E	01.1.1620E	01.1.1670E	01.1.1720E	01.1.1770E	01.802
\$9.2600E	00.8.8000E	00.8.4100E	00.8.0800E	00.8.0800E	00.8.0800E	00.802
\$7.2900E	00.7.0900E	00.6.9000E	00.6.7300E	00.6.5700E	00.6.4100E	00.802
DATA TL12/						
2.5910E	01.2.1490E	01.1.9160E	01.1.7580E	01.1.6410E	01.1.5480E	01.802
21.4720E	01.1.4070E	01.1.3510E	01.1.3030E	01.1.2200E	01.1.1530E	01.802
31.0930E	01.1.0500E	01.1.0130E	01.0.7800E	00.9.4800E	00.9.2200E	00.802
43.9700E	00.8.7500E	00.2.2480E	01.1.7340E	01.1.4980E	01.1.3660E	01.802
51.2740E	01.1.2030E	01.1.1450E	01.1.0970E	01.1.0560E	01.1.0200E	01.802
69.6000E	00.9.1100E	00.8.7200E	00.8.4000E	00.8.0900E	00.7.8400E	00.802
77.6100E	00.7.4200E	00.7.2200E	00.7.0400E	00.6.1640E	01.2.7630E	01.802
82.4650E	01.2.2420E	01.2.0730E	01.1.9440E	01.1.8420E	01.1.7580E	01.802
91.6780E	01.1.6330E	01.1.5420E	01.1.4850E	01.1.4300E	01.1.3740E	01.802
\$1.3240E	01.1.2780E	01.1.2390E	01.1.2050E	01.1.1730E	01.1.1440E	01.802
\$2.7860E	01.2.3160E	01.2.0610E	01.1.8900E	01.1.7610E	01.1.6610E	01.802
\$1.5780E	01.1.5080E	01.1.4470E	01.1.3940E	01.1.3050E	01.1.2320E	01.802


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DATA TL35/100*1./
DATA TL36/
11.196CE 00.1.245IE 00.1.290CE 00.1.2986E 00.1.2865E 00.1.2665E 00.1.2665E 00.1.2665E
21.2455E 00.1.2253E 00.1.2078E 00.1.1919E 00.1.1678E 00.1.1496E 00.1.1496E 00.1.1496E
31.1366E 00.1.1270E 00.1.1197E 00.1.1080E 00.1.1016E 00.1.0980E 00.1.0980E 00.1.0980E
41.0958E 00.1.0935E 00.1.0925E 00.1.0922E 00.1.0922E 00.1.0922E 00.1.0922E 00.1.0922E
51.0923E 00.1.0927E 00.1.0930E 00.1.0933E 00.1.0937E 00.1.0937E 00.1.0937E 00.1.0937E
61.0943E 00.1.0944E 00.1.0944E 00.1.0943E 00.1.0941E 00.1.0941E 00.1.0941E 00.1.0941E
71.0957E 00.63*1./
DATA TL40/ 95*0.0. 1.2133. 1.. 1.. 1.165. 0. /
END
BD2 551
BD2 552
BD2 553
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BD2 561-

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SUBROUTINE AGSOLN (AE,DEL,UPDOWN,AG,X)
LOGICAL ERR, FIX
SOLUTION FOR THE GEOMETRIC AREA RATIO WHEN THE EFFECTIVE AREA
RATIO AND BOUNDARY LAYER DISPLACEMENT THICKNESS ARE GIVEN

      AE = EFFECTIVE AREA RATIO
      DEL = DISPLACEMENT THICKNESS
      UPDOWN= 1. IF DOWNSTREAM SOLUTION IS DESIRED
              -1. IF UPSTREAM SOLUTION IS DESIRED
      AG = COMPUTED GEOMETRIC AREA RATIO

      COMMON /AREA/ ATPI(11,2),PARAM(3,12,2),RTHCM(2),NSECT(2),
      COMMON /NEQ/ CMDST(2),NPROFL(2),NPRFLS,NBL
      COMMON /ERROR/ ERR
      COMMON /BL/ DELBL(2),BLINT(2),XZERO,TWALL,CPWALL,VISROT,DIAM(2),
      SW,RO,JDIM,IPCINT
      DIMENSION DEL(2),DELP(2),DSTP(2)
      DATA RNAME /6HAGSOLN/
      IF (NPRFLS.EQ.2) GO TO 20
      NOT A CHANNEL
      IF (JDIM.NE.0) GO TO 10
      TWO-DIMENSIONAL FLOW
      AG=DEL(1)+CMDST(1)*AE
      GO TO 100
      AXISYMMETRIC FLOW
      SORTA=DEL(1)+CMDST(1)*SORT(AE)
      AG=SORTA**2
      GO TO 100
      CHANNEL
      DO 30 L=1,2
      DSTP(L)=RO*(1.-CMDST(L))
      DELP(L)=RO*DEL(L)
      FIX=.FALSE.
      YZC=RTHCM(1)*RTHCM(2)
      AECE=(1.-DSTP(1))/RTHCM(1))*(1.-DSTP(2))/RTHCM(2)*AE-DELP(1)*DELP(2)
      1/YOZO
      A=AE
      ICOUNT=0
      CALL FINDX (A,UPDOWN,X)
      IF (ERR) RETURN
      ICOUNT=ICOUNT+1
      IF (ICOUNT.GT.20) GO TO 90
      CALL GVAR2 (X,Y,Z)
      IF (ICOUNT.GT.1) GO TO 70
      A=AECE+(Y*DELP(2)+Z*DELP(1))/YOZO
      XO=X
      FO=A-AE
      IF (A.GE.1.) GO TO 50
      X=0.
      GO TO 60
      CALL GEOMAR (X,A,DA)
      IF (ERR) RETURN
      F=AECE+(Y*DELP(2)+Z*DELP(1))/YOZO-A
      IF (ABS(F)/A.LE.1.E-5) GO TO 120

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IF (F.EQ.FO) GO TO 120
XL=X
X=X-F*(X-XO)/(F-FO)
XO=XL
FO=F
IF (X*UPDOWN.GE.0.) GO TO 60
SIGN OF X IS INCONSISTENT WITH UPDOWN
IF (FIX) GO TO 90
IF (AE.GT.1.001) GO TO 90
TROUBLE IS DIAGNOSED AS DUE TO SONIC POINT TOO FAR DOWNSTREAM
OF GEOMETRIC THROAT. TRY FIX -- RESET DISPLACEMENT THICKNESSES
TO THEIR VALUES AT THE THROAT.
DO 80 L=1,2
DELP(L)=DSTP(L)
FIX=.TRUE.
WRITE (6,140)
GO TO 40
WRITE (6,150) AE,DEL,UPDCWN,X
NAMELIST /AGDMP/ OMDST,RO,DSTP,DELP,YOZO,RTHCM,AEC,A, AO,XO,F,FO,
WRITE (6,AGDMP)
CALL DUMP (RNAME)
GO TO 130
IF (AG.GT.1.) GO TO 110
AG=1.
X=0.
GO TO 130
CALL FINDX (AG,UPDOWN,X)
GO TO 130
AG=A
RETURN
FORMAT (29H0+++++ FIX REQUIRED IN AGSOLN)
FORMAT (30H0CONVERGENCE FAILURE IN AGSOLN,9X,3HAE=,1PE12.5,8X,4HDE
1L=.2E12.5,8X,7HUPDOWN=.0PF4.1,8X,2HX=.1PE12.5)
END

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150


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COMMON /SS/ CAS,US
COMMON /SWITCH/ XPB,DXPB,W
EQUIVALENCE (XP,XPB). (DXP,DXPB)
COMMON /DRV/ SHJGGJ
COMMON /TNCE/ SUMGH,SCPGH,QDPR,QDPE
COMMON /BLRAD/ Y070
COMMON /MODP2/ MODLPT
COMMON /CCLOSP/ CGMW,OPJ(10),IJCS(10),NCS
COMMON /TNONEQ/ CTE,DTE,BPAR,EPAR(2,25),NT,ITR(25),KTF(25),
1 KTR(25),ICH(20),IPA(25)
DATA ONE /1./, ZERO /0./
NAMELIST /DMP/ IPOINT,X,T,P,RH,U,AM,H,AE,AG,AJ,UPRIME,AS,DX,HE,HR,
1 HREF,HW,TE,TREF,COSB,DADX,YZ,DYZDX,VISC,VISCR,VISCV,
2 PR,PRREF,PRW,ALPR,ORDIN,XI,OMEGA,SIGH,DMDX,DLM,US,RO,
3 SW,NBL,IYPE,RER,BLINT,ENFRIM,SN,XSN,THETA1,THETA,HTR,
4 DELBL,DDELBL,XL,RAF,TAUW,CWDOT,XTEST,GAMMAE,WW,AVCON,B,EK1,
5 OLDMP,TRCALC,TEP,TREFP,RMR,FO,F2
DELBL=DISPLACEMENT THICKNESS
THETA=MOMENTUM THICKNESS
THETAI=MOMENTUM THICKNESS FOR TWO-DIMENSIONAL FLOW
THETAI=BOUNDARY LAYER THICKNESSES DELBL,THETA,THETA1 ARE ALL
THE BOUNDARY LAYER THICKNESSES DELBL,THETA,THETA1 ARE ALL
NONDIMENSIONALIZED USING THE GEOMETRIC THROAT RADIUS RO. THEY
ARE CONVERTED TO INCH UNITS FOR OUTPUT, IN SUBROUTINE OUT1.
IF (X.GT.XP.OR.IPOINT.EQ.1) GO TO 10
IF (X.EQ.CXR) GO TO 550
WRITE (6,DMP) X,XP
GO TO 550
10 DO 540 L=1,NPRFLS
IF (IPOINT.NE.1.OR.L.EQ.2) GO TO 90
SWP1=SW+1.
B=2.14+1.28*SWP1+0.93*(SWP1**2)**2
SM1=B-1.
EK1 AND B ARE THE COEFFICIENTS IN THE COHEN-RESHOTKO LINEAR FIT TO BLA
THE MOMENTUM PARAMETER (CAP.N) AS A FUNCTION OF THE CORRELATION
NUMBER SN (SMALL N). THE FORMULA IS N=EK1+B*SN.
SWFAC=SWP1*(7.+3.4*SWP1)
SWFAC2=1.+4.5*SWP1**0.9
HCON=CRP/CMA
HW=CPWALL*TWALL
TEQ=0.
TREFP=0.
H0=CHA#HCON
DO 20 I=1,ISS
XSAVE(I)=SAVEC(I)
XTEST(I)=-1.
DO 40 I=1,ISS
DO 30 K=1,NCS
IF (IJCS(K).NE.I) GO TO 30
SAVEC(I)=OPJ(K)
GO TO 40
CONTINUE
SAVEC(I)=0.
CONTINUE
40 CALL TRANSP (TWALL,PRESA)

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PRW=PR
VISCW=VISC
RWMW=VISCW*PRESA*CGMW/82.05/TWALL/RHAP
DO 50 I=1,ISS
SAVEC(I)=XSAVE(I)
IF (NPRFLS.EQ.1) GO TO 60
YDZO=RTHCM(1)*RTHCM(2)
ITYPE=3
GO TO 80
IF (JDIM.EQ.0) GO TO 70
ITYPE=2
GO TO 80
ITYPE=1
ITYPE IS A FLAG FOR TYPE OF NOZZLE GEOMETRY
ITYPE=1 TWO-DIMENSIONAL
ITYPE=2 AXISYMMETRIC
ITYPE=3 RECTANGULAR CHANNEL
AVCON=W/RO
RO IS SET IN MAIN
COUPLD=.FALSE.
ISMD=30
DDELBL(1)=0.
DDELBL(2)=0.
IF (L.EQ.2) GO TO 100
IF (NT.EQ.2) HO=CHA*HCON
AS=SQRT(CAS*CT)
IF (FINAL) GO TO 110
B-INTI=ELINT(L)
XII=XI(L)
IF (L.EQ.2) GO TO 220
TE=CTAP*T
DO 120 I=1,ISS
IF (ABS(XTEST(I))-SAVEC(I)).GT.0.01) GO TO 130
CONTINUE
CLDMF=.TRUE.
TRCALC=.FALSE.
GO TO 150
CLDMF=.FALSE.
DO 140 I=1,ISS
XTEST(I)=SAVEC(I)
GO TO 160
IF (ABS(TE-TEP)/TE.LT.0.005) GO TO 180
CALL TRANSP (TE,P*PRESA)
TEP=TE
IF (ERR) RETURN
IF (IPOINT.LE.1) PRREF=PR
DO 170 I=1,4
TRPSV(I)=TRP(I)
HE=HCON*H
PRPWR=PRREF*.56
HR=HE+PRPWR*(HO-HE)
HR=RECOVERY ENTHALPY
HREF=.5*(HF+HW)+.22*(HR-HE)
TREF=CTAP*HREF/HO
CALL GEOMAP (X,AG,CADIX)

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IF (ERR) RETURN
AG=GEOMETRIC AREA RATIO (AREA DIVIDED BY AREA AT THROAT)
AE=EFFECTIVE AREA RATIO ALLOWING FOR BOUNDARY LAYER DISPLACEMENT
THICKNESS
GO TO (190,200,210), ITYPE
AJ=1.
COSB(1)=1./SQRT(1.+(RO*DADX)**2)
GO TO 230
AJ=AG
COSB(1)=1./SQRT(1.+0.25*(RO*DADX)**2/AG)
GO TO 230
CALL GMAR3 (X,DYZDX(1),DYZDX(2),YZ(1),YZ(2))
LPR=LPRIME(L)
AJ=(YZ(LPR))**2
COSB(L)=1./SQRT(1.+(DYZDX(L))**2)
COSB=COSINE OF THE ANGLE BETWEEN THE NOZZLE WALL AND THE AXIS OR
CENTER PLANE.
IF (L.EQ.2) GO TO 270
UPRIME=EU*US
IF (OLDMF.AND.ABS(TREF-TRFFP)/TREF.LT.0.005) GO TO 240
CALL TRANSP (TREF,P*PRESA)
IF (ERR) RETURN
TREFP=TREF
TRCALC=.TRUE.
PRREF=PR
VISC=VISC
IF (.NOT.TRCALC) GO TO 250
VR=VISC/VRWALL
TR=TREF/TWALL
OMEGA=ALOG(VR)/ALOG(TR)
IF (IPOINT.NE.1) GO TO 250
EK1=0.38-0.76*PRREF*(1.-CMEGA)*EXP(-6.67*SWP1)
IF (EK1.GE.0.01) GO TO 250
WRITE (6,520)
WRITE (6,DMP)
EK1=0.01
DO 260 I=1,4
TRP(I)=TRPSV(I)
AM=UPRIME/AS
ORDIN(L)=AJ*AM**BM1*AS*RH*T
IF (ITYPE.EQ.2) R=SORT(AG)
R=RATIO OF LOCAL NOZZLE RADIUS TO THROAT RADIUS.
IF (IPOINT.NE.1) GO TO 290
FIRST PCINT
DX=(X-XZERO)/10.
DXSQ=DX**2
XX=XZERO
XI(L)=0.
BLINT(L)=0.
CC=AM*AG
CALL RADIUS (ITYPE,XZERO,R1,AG1,AGJ,L)
AVI=CC/AG1
DRD1=AGJ*AM1**BM1
DO 280 I=1,10
XX=XX+DX
  
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CALL RADIUS ( ITYPE,XX,R2,AG1,AGJ,L)
DXI=SQRT(DXSQ+(R2-R1)**2)
XI(L)=XI(L)+DXI
AM1=CC/AGI
ORD2=AGJ*AM1**BMI
BLINT(L)=BLINT(L)+0.5*(ORD1+ORD2)*DXI
ORD1=ORD2
R1=R2
BLINT(L)=BLINT(L)*AS
DLM(L)=-DADX/AG
X IS THE AXIAL COORDINATE DOWN THE NOZZLE, ZERO AT THE GEOMETRIC
THROAT.
XI IS THE STREAMWISE COORDINATE IN THE BOUNDARY LAYER, ZERO AT THE
NOZZLE ENTRANCE (WHERE X=XZERO).
GO TO 370
DX=X-XP
IF (INEQ.EQ.0) GO TO 320
COMPUTE DERIVATIVES FROM RATE DATA
IF (NT.EQ.1) GO TO 300
DUDX=GJ(1)*CCPJ(1)*DTE+SCPGH*DT
GO TO 310
DUDX=SCPG*DT
DUDX=-CMA/U*(DUDX+SHJDGJ)
IF (NT.EQ.2) DUDX=DUDX+DCHA/SU
DLM(L)=DUDX/U-0.5*DT/T
GO TO 360
IF (IPQINT.GT.2) GO TO 340
DMDX=(AM-AMP)/DX
GO TO 350
CALCULATE DERIVATIVE FROM QUADRATIC FIT
DX2=DX+DXP
XPXP=X+XP
DMDX=(AM-AMP)/DX
C2=(DMDX-(AM-AMPP)/DX2)/DXP
C1=DMDX-C2*XPXP
DMDX=C1+2.*C2*X
IF (DMDX.LT.0.) GO TO 330
DLM(L)=DMDX/AM
TRAPEZOIDAL RULE INTEGRATION
DXI=DX/COSB(L)
XI(L)=XI(L)+DXI
BLINT(L)=BLINT(L)+0.5*(ORDIN(L)+ORDINP(L))*DXI
DLM(L)=DLM(L)*COSB(L)
RER=RH*RHAP*UPRIME*RO/VISC
RER=REYNOLDS NUMBER BASED ON THROAT RADIUS AND FREE-STREAM
CONDITIONS.
ENPRIM=BLINT(L)/ORDIN(L)/RO*EK1
RMR=RW*W*P/VISC/RH
THETA1=SQRT(ENPRIM/RER*MR)
GO TO (380,390,380), ITYPE
THETA(L)=THETA1
GO TO 400
TRANSVERSE CURVATURE CORRECTION FOR AXISYMMETRIC NOZZLES
SQT=SQRT(1.-2.*COSB(L)*THETA1/R)

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400      THETA(L)=R/COSB(L)*(1.-SQT)
        IF (.NOT.FINAL.AND.IPOINT.NE.1) GO TO 430
        SN(L)=-ENPRIM*RO*DLM(L)
        IF (IPOINT.NE.1) GO TO 410
        XSN(L)=SN(L)
        GO TO 420
410      WW=EXP(-AVCON*DX/AG)
        XSN(L)=(1.-WW)*SN(L)+WW*XSN(L)
        XSNV=AMINI(XSN(L),ZERO)
        ALPR=ALOG(PRREF)
        GAMMAE=AS*2/(8.3144E7/CM*TE)
        IF (GAMMAE.LE.1.) GAMMAE=1.01
        SIGH=1./(1.+2./(GAMMAE-1.)/AM**2)
        HTR(L)=0.75*(EXP(4.6*XSNV))*(1.+(3.-15.6*XSNV)*SWP1)-1.)
        HTRP1(L)=(HTR(L)+1.)*(SWP1**(-0.4*ALPR)-0.47*SIGH*ALPR)
        HFPI=HTRP1(L)/T
        HF=HFPI-1.
        DELBL(L)=HF*THETA(L)
        IF (IPOINT.LE.1.OR..NOT.FINAL) GO TO 470
        IF (COUPLD) GO TO 440
        IF (IUPD.EQ.1) GO TO 470
        COUPLD=.TRUE.
        DDELBL(L)=(DELBL(L)-DELBLP(L))/(X-XP)
        IF (ISMD.EQ.1.OR.L.EQ.2) GO TO 450
        WD=1./ISMD
        GO TO 460
450      IF (L.EQ.2) GO TO 460
        WD=0.5*AMINI(ONE.(DX/DXO)**2)
        DDELBL(L)=WD*DDELBL(L)+(1.-WD)*DDELBLP(L)
        IF (ISMD.EQ.1.OR.LT.NPRFLS) GO TO 470
        ISMD=ISMD-1
        CONTINUE
        IF (FINAL) GO TO 480
        BLINT(L)=BLINTI
        XI(L)=XII
        GO TO 540
480      DELBLP(L)=DELBL(L)
        DDELBLP(L)=DDELBL(L)
        OM1=AMAX1(1.-OMEGA,ZERO)
        F3=EXP(-0.427*(SWP1**(-OM1))-1.))
        AX=15.5*EXP(-29.*SWP1)+0.67*EXP(4.37*SWP1)
        IX=0.7+0.47*SWP1
        F1=0.76*OM1**1.2*(1.-SORT(1.-SIGH))
        IF (SN(L).LT.-1.E-20) F1=F1/(1.+AX*(-SN(L))**BX)
        IF (FC) 490,490,500
        F3=0.
        GO TO 510
490      F3=EXP(3.5*AMINI(SN(L),ZERO)/FO**1.57)
        F2=FO+0.45*OM1*(1.-F3)+F1
        IF (SN(L).LE.-1.E-20) GO TO 520
        XL=C.2205
        GO TO 530
500      XL=0.2205+SWFAC2/(-2.7/SN(L)+5.06/(-SN(L))**0.224)
        XL=F2*XL
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520
530

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SUBROUTINE BLCALL (FINAL)
CALLING ROUTINE FOR BOUNDARY LAYER IN NCNEEQUILIBRIUM CALCULATIONS
LOGICAL FINAL
REAL ACOM(30),ELMENT(10),HP(20)
DOUBLE PRECISION AA,AAA,CAPX,GJ,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A, AFNTS, AFNX, AMACH, AR, ARBA, ARBB, BZERO,
C, CARB, CH, CHT, CLNT, CM, CMA, CRA, CRP,
1 CRRB, CRS, CSTA, CTA, CTAD, CTB, CTC, CTMAX, CTMXX,
2 CTP, CTPL, CTT, CX, CXB, CXMAX, DATEST, DBTEST, DELT1,
3 DELT2, DELTAX, DLOGA, DLOGR, DT, ENT, FLUX, HDELX, PCT,
4 DELT2, DELTAX, DLOGA, DLOGR, DT, ENT, FLUX, HDELX, PCT,
5 PCTEST, PRESA, PRESB, PRESTH, PRHC, RHAP, RHOB, RHOB,
6 RHOBAR, RHOC, RHOP, RHPL, RHTH, ROBAR, ROBARP, SCPG, SOT,
7 SEN, SHPG, SHP, SL, SL64, SM, SU, SU2, SUMG,
8 TEST, TESTB, TPRINT, TSTOP, UP, ZPA
COMMON BE(64), BET(20), BLBK(31), CAI(64), CAPXTH(20), BLC
1 CCPJ(20), CEACT(64), CGI(20), CGMU(20), CHII(20), BLC
2 CLNIMC(64), CLNPI(64), CMW(20), ETAI(64), ETAJ(20), BLC
3 GJ5(20), PERTGJ(20), PGJ(20), PI(64), PICH(64), GM(20), BLC
4 OUI(64), SAJ(20), SDCHI(64), SENT(20), SHJ(20), SHJA(20), BLC
5 SKIL(20), SS(20), TB(30), TFA(20), TFB(20), TFC(20), BLC
6 TFD(20), TFE(20), TFK(20), THEV(20), XMJAT(20), XNUI(64)
COMMON BETA(64,20), ELJ(10,20), GELJ(10,20), BLC
1 XNUIJ(64,20), XNUIJP(64,20)
COMMON IC, IM, INEQ, INEQV, IP, IRUN, ISC, ISCP1,
1 ISMC, ISMCK, ISR, ISS, ISSNR, ISSP2, ISSP3, ISSP4,
2 ISW1A, ISW1B, ISW2A, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
3 ISW5B, ISW6A, ISW6B, IUPD, IZERD, JJK, LC, MI, NFIT,
4 NIT, NNN, NNS, NOS, NOT, NTEST
COMMON IGM(20), ITB(5), KUR(64,20), LPIJ(20,10)
COMMON ACOM, ELMENT, HP
DIMENSION AAA(22,24), BTA(64,20), CAPO(31), CCI(20), DGJ(20), SBRLC
1J(20), SDGJ(20), SHJAP(20), THEVP(20)
EQUIVALENCE (AA(1,1), AAA(1,1)), (BETA(1,1), BTA(1,1)),
1 (BLHK(1), CAPO(1)), (CAPXTH(1), CCI(1)), (GJA(1), DGJ(1)),
2 (CAPX(1), GJ(1)), (SAJ(1), SBJ(1)), (SS(1), SDGJ(1)),
3 (SHJA(1), SHJAP(1)), (THEV(1), THEVP(1))
COMMON /TEMPRY/ SAVEC(20)
IF (INEQ.EQ.0) GO TO 20
DO 10 J=1,ISS
SAVEC(J)=GJ(J)*CM
GO TO 40
CT=CT+PCT
RHO=RHO+PRHO
DO 30 J=1,ISS
SAVEC(J)=(GJ(J)+PERTGJ(J))*CM
IF (ISW3B.EQ.0) GO TO 50
CALL BLAYER (FINAL)
IF (INEQ.NE.0) GO TO 60
CT=CT-PCT
RHO=RHO-PRHO
RETURN
END

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C

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SUBROUTINE BXSECT (L,MV,I,J)
SUBROUTINE BXSECT DETERMINES THE CORRESPONDENCE BETWEEN THE KQ ANDBXS
V ARRAYS AND SEARCHES THE V ARRAY FOR REFERENCES TO SPECIES PAIRS. BXS
COMMON /TRANS7/ V(400),KQ(100),NQ(100),IQ(400),JO(400),NKQ
COMMON /TRANS8/ NV(15),N
IF (L.NE.0) GO TO 40
MV=1
MADD=0
L=L+1
K=KQ(L)
IF (K.EQ.9) GO TO 30
IF (K.NE.12) GO TO 50
MADD=4
MM=MV+MADD
I=V(MM+1)
J=V(MM+2)
RETURN
V(MM+1)=I
V(MM+2)=J
IF (MADD.NE.0) GO TO 20
MV=MV+NV(K)
IF (L.LT.NKQ) GO TO 10
L=0
RETURN
END

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SUBROUTINE COMM
LOGICAL ERR, FAILED, LOGC, NOREAC, RCALC
REAL ACOM(30), ELEMENT(10), HP(20)
DOUBLE PRECISION AA, AAA, CAPX, GJ, CDIJ
COMMON AA(22,24), CDIJ(20,10), CAPX(20)
COMMON A, AFNTS, AFNH, AMACH, AR, ARBA, ARBB, BZERO,
C, CARB, CH, CHA, CTAP, CTB, CMA, CRA, CRP,
1 CRB, CRS, CSTA, CTT, CX, CXB, CXMAX, DATEST, DBTEST, DELT1,
2 CTP, CTPL, CLOGA, DLOGR, PRESTH, PRHO, RHAP, RHOR,
3 DELT2, DELTAX, DLOGA, PRESA, RHOP, RHTH, ROBAR, ROARP, SDT,
SPTTEST, PRES, RHOC, RHOP, RHP, RHTH, ROBAR, ROARP, SPCG,
6RHOBAR, RHOC, RHOP, RHP, RHTH, ROBAR, ROARP, SU2,
7 SEN, SHPG, SC, SL, SL04, SM, SU, SUMG,
8 TEST, TESTB, TPRINT, TSTOP, UP, ZPA,
COMMON BE(64), BET(20), BLBK(31), CAI(64), CAPXTH(20),
1 CCPJ(20), CFACT(64), CGI(20), CGMU(20), CHI(64), CHI(20),
2 CLNIMC(64), CLNPI(64), CMW(20), ETAI(54), ETAJ(20),
3 GJJ(20), PERTGJ(20), PGJ(20), PI(64), PICH(64),
4 Q(64), SAJ(20), SDCHI(64), JENT(20), SHJ(20),
5 SKIL(20), SS(20), TH(30), TFA(20), SHJA(20),
6 TFD(20), TFE(20), TFK(20), THEV(20), XMJAT(20),
COMMON BETA(64,20), ELJ(10,20), GELJ(10,20),
1 XNUIJ(64,20), XNUIJP(64,20)
COMMON IC, IM, INEQ, INFOV, IP, IRUN, ISC, ISCP1,
COMMON ISMC, ISMCNR, ISR, ISS, ISSNR, ISSP1, ISSP2, ISSP3, ISSP4,
1 ISW1A, ISW1B, ISW2A, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
2 ISW5B, ISW6A, ISW6B, IUPD, IZERO, JJK, LC, MI, NFIT,
3 NIT, NNN, NNS, NOS, NQT, NTEST
COMMON IGJ(20), IGM(20), ITB(5), KUR(64,20), LPIJ(20,10)
COMMON ACOM, ELEMENT, HP
DIMENSION AAA(22,24), BTA(64,20), CAPO(31), CCI(20), DGJ(20), GJ(20), SB
1J(20), SDGJ(20), SHJAP(20), THEVP(20)
EQUIVALENCE (AA(1,1), AAA(1,1)), (BETA(1,1), BTA(1,1)),
1 (BLBK(1), CAPO(1)), (CAPXTH(1), CCI(1)), (GJA(1), DGJ(1)),
2 (CAPX(1), GJ(1)), (SAJ(1), SBJ(1)), (SS(1), SDGJ(1)),
3 (SHJA(1), SHJAP(1)), (THEV(1), THEVP(1))
COMMON /ERROR/ ERR
COMMON /CERROR/ FAILED
COMMON /TNEG/ CTE, DTE, PPAR, EPAR(2,25), NT, ITR(25), KTF(25),
1 KTR(25), ICH(20), IPA(25)
COMMON /AREA/ ATPI(11,2), PARAM(3,12,2), RTHCM(2), NSECT(2),
1 NSECTU(2), ISHAPE(12,2), NPROFL(2), NPROFLS, NBL
COMMON /TNDG/ XMJATD(20,2), CLNTD(2), CTD(2)
COMMON /TNCE/ SUMGH, SPCGH, ODPR, ODPE
COMMON /GLIM2/ NOREAC(64)
DATA VIC1, VIC2 /1, 094E24, 1, 1328E-16/, VNC /4, 99E29/
DIMENSION ALGJ(20)
EQUIVALENCE (ALGJ(1), AA(1,24))
NAMELIST /CCMDMP/ CX, AFNX, DLOGA, RHO, DLOGR, SUMG, SHPG, RHAP, CRP,
1 SU2, CSTA, Z1, Z2F, CLNTD, GJ, CHI, PI, XMJATD, CTD, SUMGH, SPCGH,
2 ODPR, ODPE, RKR, RKF, UELAS, CLNPI, CLNIMC
DATA ICOUNT /0/, ONE /1./
DATA ICYCLE /90000000/
ICOUNT=ICOUNT+1
IF (ISW5B.LT.0) ICYCLE=-ISW5B

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10 IF (ICYCLE.EQ.90000000) GO TO 30
11 IF (MOD(ICOUNT,ICYCLE)) 10.20.10
12 J=J+1
13 ISWB=1
14 WRITE (6,550) ICOUNT
15 CONTINUE
16 FAILED=.FALSE.
17 SUMG=0.
18 SCPR=0.
19 QDPR=0.
20 QDPE=0.
21 DO 70 J=1,ISS
22 IF (GJ(J)) 80.40.50
23 ALGJ(J)=-1000.
24 GO TO 60
25 ALGJ(J)=ALOG(GJ(J))
26 IF (J.LT.NT) GO TO 70
27 SUMG=SUMG+GJ(J)
28 SCPR=SCPR+GJ(J)*CCPJ(J)
29 CONTINUE
30 SUMGH=SUMG
31 SCPG=SCPR
32 IF (NT.EQ.1) GO TO 100
33 SUMG=SUMG+GJ(1)
34 SCPG=SCPG+GJ(1)*CCPJ(1)
35 TEPECTE*CTAP
36 GO TO 100
37 WRITE (6,560)
38 FAILED=.TRUE.
39 RETURN
40 IF (INEQ) 110.150.110
41 SFN=0.0
42 SHPG=0.
43 DO 130 J=1,ISS
44 IF (GJ(J)) 80.130.120
45 SEN=SEN+CJ(J)*((SENT(J)-ALGJ(J))
46 SHPG=SHPG+GJ(J)*SHJ(J)
47 CM=1./SUMG
48 CHECMA*SHPG
49 SU2=2.*((CHA-CH)
50 IF (SU2.GE.0.) GO TO 140
51 WRITE (6,570)
52 GO TO 90
53 SU=SORT(SU2)
54 CALL GECM
55 IF (ERR) RETURN
56 RHOP=RHOP+RHAP
57 RHPL=ALOG(RHOP)
58 Z1=ALOG(SL64/SU)+CSTA
59 Z2F=41.2918*CRP
60 RCALC=.FALSE.
61 MAIN LCP OVER THE REACTIONS
62 DO 460 I=1,ISR
63 IF (INCREAC(I)) GO TO 450

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COMPUTATION OF PICHI
LOGC=.FALSE.
IF (NT.EQ.1) GO TO 160
KF=KTF(I)
KR=KTR(I)
GO TO 170
160 KF=1
KR=1
IF (KF-3) 180,190,200
CLKF=ETAI(I)*CLNTD(KF)+CAI(I)-CEACT(I)/(CTD(KF)*CRP)
GO TO 240
190 CLKF=ETAI(I)*CLNTD(2)+CAI(I)+ALOG(1.-EXP(-CEACT(I)/(CTD(I)*CRP)))
GO TO 240
200 IF (RCALC) GO TO 220
RCALC=.TRUE.
IF (NPRFLS.EQ.2) GO TO 210
CALL GMAR (CX,R)
GO TO 220
210 CALL GMAR2 (CX,Y,Z)
R=1./SORT(0.5*(1./Y**2+1./Z**2))
IF (KF.FQ.4) GO TO 230
CLKF=CAI(I)-0.5*ALOG(R)
GO TO 240
230 IPAI=IPA(I)
TAU=SPAR*RHOP*GJ(IPAI)*R
CLKF=ETAI(I)*CLNTD(2)-ALOG(AMAX1(ONE,TAU))+CAI(I)
CLNPI(I)=CLKF+ZI
CLKR=CLKF
CLNIMC(I)=0.0
T6=0.0
T7=0.0
IF (KR.EQ.0) GO TO 250
Z2=Z2F*CTD(KR)
Z3=Z2*RHOP
CLZ2=ALOG(Z2)
CLZ3=ALOG(Z3)
DD 280 J=1,ISS
IF (XNUI(J,I,J).EQ.0.0) GO TO 260
T6=T6+XNUI(I,J)*ALGJ(J)
IF (KUR(I,J).EQ.0) GO TO 270
T7=T7+GJ(J)
IF (BETA(I,J).EQ.0.0 OR KR.EQ.0) GO TO 280
CLNIMC(I)=CLNIMC(I)+BETA(I,J)*(ALGJ(J)+CLZ3+XMJATD(J,KR))
CONTINUE
IF (KR.EQ.0) GO TO 300
IF (CLNIMC(I).LT.20.) GO TO 290
LOGC=.TRUE.
GO TO 310
290 CHI(I)=1.0-EXP(CLNIMC(I))
GO TO 310
300 CHI(I)=1.0
310 CLNPI(I)=CLNPI(I)+(XNUI(I)-1.0)*RHPL+T6
IF (QQ(I)) 320,350,320
IF (T7) 330,340,330
330 CLTBF=ALOG(RHOP*T7)
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340 GO TO 350
    PI(I)=J.0
    CLTBF=-10000000.
    GO TO 370
350 CLTBF=0.
360 PI(I)=EXP(CLNPI(I))+CLTBF)
370 IF (LOGC) GO TO 380
    PICH(I)=PI(I)*CHI(I)
    GO TO 390
380 PICH(I)=-EXP(CLN1MC(I))+CLNPI(I)+CLTBF)
390 IF (NT.EQ.1) GO TO 460
C COMPUTATION OF RADIATIVE LOSS QDPR AND ENERGY TRANSFER TO THE
C ELECTRON GAS QDPE
    SUMF=0.
    RKF=EXP(CLKF)
    SUMR=0.
    CALL EPART (I,TEP,EF,ER,OF,OR)
    DO 410 J=1,ISS
    IF (KR.EQ.0) GO TO 400
    CLKR=CLKR+RETA(I,J)*(CLZ2+XJATD(J,KR))
    SUMF=SUMF+XNUIJ(I,J)*(RHPL+ALGJ(J))
    SUMR=SUMR+XNUIJP(I,J)*(RHPL+ALGJ(J))
    IF (KR) 420,420,430
400 RKR=0.
410 GO TO 440
420 RKF=EXP(CLKF)
430 SUMF=EXP(SUMF+CLTBF)
440 SUMR=EXP(SUMR+CLTBF)
    QDPR=VDPR+OF*RKF*SUMF-QR*RRKR*SUMR
    QDPE=QDPE+EF*RKF*SUMF-ER*RRKR*SUMR
    NAMELIST /CDP/ I,KF,KR,CLKF,CLKR,T6,T7,Z2,Z3,CLZ2,CLZ3,
1 CLTBF,RKF,RRKR,EF,OF,OR,SUMF,SUMR,GDPR,QDPE
    IF (INEQ.NE.0.AND.ISWEB.NE.0) WRITE (6,CDP)
    GO TO 460
450 PICH(I)=0.
460 CONTINUE
470 IF (INEQ) 470,540,470
480 PRES=CT*SUMG
490 PRES=CT*SUMGH+CTE*GJ(1)
C COMPUTATION OF ENERGY TRANSFER TO THE ELECTRON GAS BY ELASTIC
C COLLISIONS
    TEP=TEP**3
    VIC=VIC1*RHOP/SQRT(TEQ)*ALOG(VIC2*TEQ/RHOP/GJ(1))
    OELAS=0.
    VNCC=VNC*SQRT(TEP)*RHOP*PIOMEG(TEP)
    DO 520 J=2,ISS
    ICHG=ICH(J)
    GO TO (500,510), ICHG
    VEJ=VNCC*GJ(J)
    GO TO 520
500 VEJ=VIC*GJ(J)
510 OELAS=OELAS+VEJ/CGI(J)
520 OELAS=3.*CRA*RHOP*CGI(1)*CTAP*(CT-CTE)*GJ(1)*OELAS

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530 QDPE=QDPE+QELAS
540 PRES=PRES+RHO*CMA/ROBARA
540 SEN=CRA*(SEN-ALOG(PRES*CM*PRESA)/CM)
540 CONTINUE
540 IF (INEQ.NE.0.AND.ISW5B.NE.0) WRITE (6,COMDMP)
540 RETURN
550 FORMAT (//11H **ICOUNT=.15)
560 FORMAT (43HNEGATIVE CONCENTRATION ENCOUNTERED IN COMM)
570 FORMAT (24H0SU2 LESS THAN 0 IN COMM)
570 END

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SUBROUTINE CXSECT (L,J)
 SUBROUTINE CXSECT DETERMINES THE CORRESPONDENCE OF SPECIES PAIRS
 BETWEEN THE MASTER SPECIES LIST AND THE TRANSPORT CALCULATIONS.
 COMMON /TRANS2/ KKQ(100),NNG(100),ISEQ(100),NNKQ,I(50)

M=L-J
 L=I(L)
 J=I(J)
 IF (ISIGN(M,L-J).EQ.M) GO TO 10
 M=L
 L=J
 J=M
 RETURN
 END

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1 SUBROUTINE DERIVS
2 LOGICAL ERR,FAILED
3 REAL ACOM(30),ELEMENT(10),HP(20)
4 DOUBLE PRECISION AA,CAPX,CDIJ
5 COMMON AA(22,24),CDIJ(20,10),CAPX(20)
6 COMMON A,AFNLS,AFNX,AMACH,AR,ARBA,ARBB,BZERO,
7 C,CRB,CRS,CSTA,CT,CTAP,CTB,CMA,CRA,CRP,
8 CTP,CTPL,CTT,CX,CXB,CXMAX,DATEST,DBTEST,DELT1,
9 DELT2,DELTA,DLOGA,DLOGR,DLOGS,PRESTH,PRHC,PRH,PHOB,
10 SPCTEST,PRES,PRESA,RHOC,RHOP,RHPL,RHTR,RORARA,ROBARP,SCPG,SDT,
11 GRHOBAR,RHOC,RHOP,RHPL,RHTR,RORARA,ROBARP,SCPG,SDT,
12 SEN,SHPG,SC,SL,SL64,SM,SU,SUMG,
13 TEST,TESTB,TPRINT,TSTOP,ZP,ZPA,
14 COMMON BE(64),BET(20),BLBK(31),BLM(64),CAPXTH(20),
15 CCPJ(20),CEACT(64),CGI(20),CGMU(20),CHI(64),CHII(20),
16 CLN1MC(64),CLNFI(64),CMW(20),ETAI(64),ETAJ(20),
17 GJ8(20),PERTGJ(20),PGJ(20),PI(64),PICH(64),
18 Q(64),SAJ(20),SDCHI(64),SENT(20),SHJ(20),
19 SKIL(20),SS(20),TFC(20),TFB(20),TFE(20),TFM(64),
20 TFD(20),TFK(20),THEV(20),XNJAT(20),XNUI(64),
21 COMMON BETA(64,20),ELJ(10,20),GELJ(10,20),
22 XNUIJ(64,20),XNUIJP(64,20)
23 COMMON IC,IM,INEQ,INEQV,IP,IRUN,ISC,ISCP1,
24 ISMC,ISMCNR,ISR,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
25 ISW1A,ISW1B,ISW2A,ISW2B,ISW3A,ISW3B,ISW4A,ISW4B,ISW5A,
26 ISW5B,ISW6A,ISW6B,IUPD,IZERO,JJK,LC,MI,NFIT,
27 NIT,NNN,NNS,NQT,NTEST,KUR(64,20),LPIJ(20,10),
28 COMMON IGJ(20),IGM(20),ITB(5),
29 COMMON ACOM,ELEMENT,HP,INEQ,INEQV,IP,IRUN,ISC,ISCP1,
30 COMMON /BL/DELBL(2),BLINT(2),XZERO,TWALL,CPWALL,VISROT,DIAM(2),
31 SW,RO,JDIM,IPCINT
32 COMMON /ERROR/ERR
33 COMMON /AREA/ATPI(11,2),PARAM(3,12,2),RTHCM(2),NSECT(2),
34 NSECTU(2),ISHAPE(12,2),NPROFL(2),NPRELS,NBL
35 COMMON /TONEQ/CTE,DTE,BPAR,EPAR(2,25),NT,ITR(25),KTF(25),
36 KTR(25),ICH(20),IPA(25)
37 COMMON /TNDG/XMJATD(20,2),CLNTD(2),CTD(2)
38 DIMENSION DELI1(2),DELI2(2),DELO1(2),DELO2(2)
39 NAMELIST /DRVDMP/ ITRY,DELBL,DELI1,DELI2,DELO1,DELO2,DEN
40 ITRY=0
41 IF (NT.EQ.1) GO TO 30
42 CTSAVE=CT
43 CT=CTE
44 CALL THERM
45 CLNTD(2)=CLNT
46 HJE=SHJ(1)
47 CPJE=CCPJ(1)
48 SENTE=SENT(1)
49 DO 20 I=1,ISS
50 XMJATD(I,2)=XMJAT(I)
51 CT=CTSAVE
52 CALL THERM
53 CLNTD(1)=CLNT
54
55
```

10

20

30

56 DER
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40 CTD(1)=CT
    DO 40 I=1,ISS
      XMJATD(I,1)=XMJAT(I)
      IF (NT.EQ.1) GO TO 50
      SHJ(I)=HJE
      CCPJ(I)=CPJE
      SENT(1)=SENTE
      CTD(2)=CTE
50 CALL COMM
      IF (ERR.OR.FAILED) RETURN
      CALL EXACT
      IF (ERR) RETURN
      IF (DLGGA*CX.GE.-0.01) GO TO 60
      FAILED=.TRUE.
      RETURN
60 IF (ISW38.EQ.0.CR.IUPD.EQ.1) GO TO 140
      DO 70 K=1,NPRFLS
        DELI2(K)=DELI1(K)
        DELI1(K)=DELBL(K)
      CALL BLCALL(,FALSE.)
70 IF (ERR) RETURN
      IF (ABS(1.-DELBL(1)/DELI1(1)).GT.0.01) GO TO 80
      IF (NPRFLS.EQ.1) GO TO 130
      IF (ABS(1.-DELBL(2)/DELI1(2)).LE.0.01) GO TO 130
80 DO 90 K=1,NPRFLS
        DELO2(K)=DELO1(K)
        DELO1(K)=DELBL(K)
90 ITRY=ITRY+1
      IF (ITRY.EQ.1) GO TO 10
      IF (ITRY.GT.6) GO TO 110
      DO 100 K=1,NPRFLS
        DEN=DELO2(K)-DELO1(K)+DELI1(K)
        IF (DEN.EQ.0.) GO TO 120
100 DELBL(K)=(DELI1(K)*DELO2(K)-DELO1(K)*DELI2(K))/DEN
        GO TO 10
110 WRITE (6,150)
120 *WRITE (6,DRVDMP)
130 CONTINUE
140 RETURN
      C
      C
      C
150 FORMAT (39H BOUNDARY LAYER ITERATION NOT CONVERGED)
      END
  
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SUBROUTINE DSMSCL (A,I,N)
DOUBLE PRECISION A
BASED ON SUBROUTINE SIMQ FROM IBM SCIENTIFIC SUBROUTINE PACKAGE
DATA RNAME /6HDSMSOL/
DIMENSION A(22,24)
I=I+1
M=I+1
L=0
DO 20 K=1,I
DO 10 J=1,I
LN=L/N
L=L+1
LM=L-N*LN
LNI=LN+1
A(LM,LNI)=A(J,K)
IF (K.EQ.1) A(J,NI)=A(J,M1)
CONTINUE
CONTINUE
CALL SIMQ (A(1,1),A(1,NI),I,KS)
IF (KS.EQ.0) GO TO 30
WRITE (6,50)
CALL DUMP (RNAME)
RETURN
DO 40 J=1,I
A(J,M1)=A(J,NI)
RETURN

```

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30
40
C
50

FORMAT (36H0MATRIX OF COEFFICIENTS IS SINGULAR.)
END

SUBROUTINE DUMP (RNAME)
LOGICAL ERR
COMMON/ERROR/ ERR
ERR=TRUE
WRITE (6,10) RNAME
RETURN

FORMAT (11H1,24H DUMP ROUTINE CALLED BY ,A6)
END

1 DUM
2 DUM
3 DUM
4 DUM
5 DUM
6 DUM
7 DUM
8 DUM
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C
10

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SUBROUTINE DUMPEX
LOGICAL DATAE,FAILED,COUPLD,NOREAC,MODLPT
DOUBLE PRECISION AA,CAPX,CDIJ,AIN
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A,AFNTS,AFNX,AMACH,AR,ARBA,ARBB,BZERO,
1 CRRB,CRS,CSTA,CTA,CTAP,CTB,CMA,CRA,CRP,
2 CTP,CTPL,CTT,CX,CTXB,CXMAX,DATEST,DBTEST,DELT1,
3 DELT2,DELTAX,DLOGA,DLOGR,PRESTH,PRHC,PRHAP,RHO,
4 SPTTEST,PRES,PHUC,RHOP,SC,SL,SL64,SM,SU,SDT,
5 SEN,SHPG,SHR,SC,SL,SL64,SM,SU,SDT,SUMG,
6 TEST,TESTB,TPRINT,TSTOP,UP,ZP,ZPA,
COMMON RE(64),BLBK(31),CAI(64),CAPXTH(20),
1 CCPJ(20),CEACT(64),CGI(20),CGMU(20),CHI(64),
2 CLNIMC(64),CLNPI(64),CMW(20),ETAI(64),ETAJ(20),
3 GJB(20),PERTGJ(20),PGJ(20),PI(64),PICH(64),
4 QQ(64),SAJ(20),SDCHI(64),SENT(20),SHJ(20),
5 SKIL(20),SS(20),TBJ(30),TFA(20),TFB(20),
6 TFD(20),TFK(20),THEV(20),XMJAT(20),
COMMON BETA(64,20),ELJ(10,20),GELJ(10,20),
1 XNUIJ(64,20),XNUIJP(64,20),
COMMON IC,INEQ,INEQV,IP,IRUN,ISC,ISCP1,
1 ISMC,ISMENR,ISR,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
2 ISW1A,ISW1B,ISW2A,ISW2B,ISW3A,ISW3B,ISW4A,ISW4B,ISW5A,
3 ISW5B,ISW6A,ISW6B,IURD,IZERO,JJK,LC,
4 NIT,NNN,NNS,NQS,NOT,NTEST,M1,NFIT,
COMMON IGJ(20),IGM(20),ITB(5),KUR(64,20),LPIJ(20,10),
COMMON TEMPRY/SAVEC(20),
COMMON /BL/DELBL(2),BLINT(2),XZERO,TWALL,CPWALL,VISROT,DIAM(2),
1 SW,RC,JDIM,IPCINT,
COMMON /TRPROP/VISC,PR,SIGMA,FLEWIS
COMMON /NEQ/OMDST(2),DDELBL(2)
COMMON /AREA/ATPI(11,2),PARAM(3,12,2),RTHCM(2),NSECT(2),
1 NSECTU(2),ISHAPE(12,2),NPROFL(2),NPRELS,NBL
COMMON /EOC/ZCAP(20),ZSEN,ZCH,ZCM,ZRBP,ZRHO
COMMON /STAG/TS,EPSLON,TSF,EPSE
COMMON /MCDP2/ MODLPT
COMMON /MODPAR/ XMPI,DXMP,FSTAG,CATFAC,TMDEL,XMODP1,DXMODP,
1 TPLATE,KDIM
COMMON /NEWMP/ FACMP,NMDOPT
COMMON /VODPT/ TSDIAM(20),TSAR(20),NTS,MBL
COMMON /CERROR/ FAILED
COMMON /GLIM2/ NOREAC(64)
COMMON /BLNF/ XI(2),ORDINP(2),DELBLP(2),DDBLP(2),AMP,AMP,
1 COUPLD,ISMD
COMMON /DRV/ SHJGGJ
COMMON /BLOUT/ REPF,THETA(2),SN(2),XSN(2),PRREF,HR,C/DOT(2),
1 TAUW(2)
COMMON /SWITCH/ XPB,DXPB,W
COMMON /CONVRT/ CF(6)
COMMON /TAPOUT/ XXX(10),ITPOUT,NRCOUT,IFLOW,ITYPER,IMP,DATAE
COMMON /AEGECM/ SORTA,S1,S2
COMMON /THRT/ RSA
COMMON /TNONEQ/ CTE,DTE,BPAR,EPAR(2,25),NT,ITR(25),KTF(25),

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1 KTR(25), ICH(20), IPA(25)
COMMON /TNDK/ XMJATD(20,2), CLNTD(2), CTD(2)
COMMON /TNERK/ SDTE, CTEB, OEMAX
COMMON /TNCE/ SUMGH, SCPGH, ODPR, QDPE
COMMON /INGNE/ AIN(10,10)
NAMELIST /DMP1/
1 A, AFNTS, AFNX, AMACH, AR, ARBA, ARBB, BZERO,
2 C, CARB, CH, CHA, CM, CRA, CRP,
3 CRRB, CRS, CSTA, CT, CTAP, CTB, CTMAX, CTMXX,
4 CTP, CTPL, CTT, CX, CXB, CXMAX, DATEST, DBTEST, DELT1,
5 DELT2, DELTAX, DLOGA, DLOGR, DRESTH, ENT, FLUX, HDELX, PCT,
6 PCTEST, PRES, PRESA, PRHOC, PRHAP, PRHG, RHOB, RHO,
7 RHOBAR, RHOC, RHOP, RHPL, RHTH, ROBAR, ROBARP, SCPG, SDT,
8 SEN, SHPG, SHC, SL, SL64, SM, SU, SUMG,
9 TEST, TESTB, TPRINT, TSTOP, UP, ZP, ZPA
NAMELIST /DMP2/
1 BE, BET, BLBK, CAI, CAPX, CAPXTH, CCPJ,
2 CEACT, CGI, CGMU, CHI, CHII, CLNIMC, CLNFI, C'W, ETAI, ETAJ, GJA,
3 GJB, PERTGJ, PGJ, PI, PICH, QM, QG, SAJ, SCCHI, SENT, SHJ,
4 SHJA, SKIL, SS, TB, TFA, TFB, TFC, TFD, TFE, TFK, THEV, XMJAT, XNUI
NAMELIST /DMP3/
1 IC, INEQ, INEGV, IP, IRUN, ISC, ISCP1,
2 ISMC, ISMCNR, ISR, ISS, ISSNR, ISSP1, ISSP2, ISSP3, ISSP4,
3 ISW1A, ISW1B, ISW2A, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
4 ISW5B, ISW6A, ISW6B, IUPD, IZERO, JJK, LC, MI, NFIT,
5 NIT, NNN, NNS, NOS, NOT, NTEST, IGJ, IGM, ITB
NAMELIST /DMP4/ DELBL, BLINT, TWALL, CPWALL, VISROT, RO,
1 DIAM, SW, JDIM, IPOINT, VISC, SAVEC, XZERO, P'REF, HR,
2 QMDST, DDELBL, ZCAP, ZSEN, ZCH, ZCM, ZRBP, ZRHO, TS, EPSLON,
3 TSF, EPSF, THETA, SN, XSN, REPF, QWDCI, TAUW, SHJ, DGJ,
4 XPB, DXPR, CF, XMP1, FSTAG, CATFAC, MODEL, XMODP1,
5 KDIM, AIP, PARAM, RTHCM, NSECT, ISHAPE, PR, SIGMA,
6 FLEXIS, FACMP, NMODPT, TSDIAM, NTS, TSAR, NPROFL, NPRELS, NBL, SORTA,
7 S1, S2, RSA, W, SUMGH, SCPGH, ODPR, ODPE, SCTE, CTEB, CTE, DTE, BPAR,
8 EPAR, NT, ITR, KTF, KTR, ICH, IPA, XMJATD, CLNTD, CTD, OEMAX, AIN,
9 FAILED, NOREAC, XI, URDINP, DELHLP, ODBLP, AMPP, AMP, COUPLD, ISMD
WRITE (6, DMP1)
WRITE (6, DMP2)
WRITE (6, DMP3)
WRITE (6, DMP4)
IF (.NOT. DATAE) GO TO 20
DO 10 I=1, NRRCOUT
BACKSPACE ITPOUT
WRITE (6, 30) ITPOUT, NRRCOUT
NRRCOUT=0
WRITE (6, 4C)
RETURN
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C
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END

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FORMAT (11H0*****TAPE, I3, 14H BACKSPACED BY, I4, 41H RECORDS TO DELETED
ITE DATA FROM FAILED CASE.)
FORMAT (1H1)

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ELC

SUBROUTINE ELCOND (SIGMA,SSIG)
 SUBROUTINE ELCOND COMPUTES ELECTRICAL CONDUCTIVITY IN MHOS/CM.
 COMMON /TRANS1/ T,Q(3,20,20),ZM2(20)
 COMMON /TRANS3/ B(20,2),BR(20,2),A(2),X(20),DH(20)
 COMMON /TRANS5/ N,IELEC,IDI,ID2
 SIGMA=0.
 IF (IELEC.EQ.0) GO TO 20
 DO 10 I=2,N
 SIGMA=SIGMA+X(I)*Q(I,1,I)
 SIGMA=SSIG*X(I)/SIGMA
 RETURN
 END

C

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SUBROUTINE EQCALC (T,P)
 LOGICAL ERR
 CALCULATES THERMOCHEMICAL EQUILIBRIUM FOR SPECIFIED T AND P.
 T=TEMPERATURE IN DEG. K
 P=PRESSURE IN ATM.

REAL ACCM(30),ELEMENT(10),HP(20)
 DOUBLE PRECISION AA,AAA,CAPX,GJ,CDIJ
 COMMON AA(2,24),CDIJ(20,10),CAPX(20)

COMMON A, AFNTS, AFNX, AMACH,
 C, CARB, CH, CHA, CLNT,
 CRB, CRS, CSTA, CT, CTAP,
 CTP, CTP, CTT, CX, CXR, CXMAX,DATEST,DBTEST,DELTX,
 DELT2,DELTX, DLOGR, DT, ENT, FLUX, HDELX, PCT,
 SPCTEST, PRES, PRESA, PRESB,PRESTH, PRMC, RNAP, RHO, RHOB,
 RHOBAR, RHOC, RHOP, RHPL, RHTH,ROBAR,ROARP, SCRG, SOT,
 SEN, SHPG, SC, SL, SL54, SM, SU, SU2, SUMG,
 TEST, TESTB, TPRINT, TSTOP, UP,
 ZP, ZPA

COMMON BE(64), BLBK(31), BLBK(64),CAPXTH(20),EQC
 CCPJ(20), CEACT(64), BET(20), CGMU(20), CHI(54), CHI(20),EQC
 CLNIMC(64), CLNPI(64), CMW(20), ETAI(64), ETAJ(20),EQC
 GJB(20),PERTGJ(20), PGJ(20), PI(64), PICH(64), QM(20),EQC
 QQ(64), SAJ(20), SDCHI(64), SENT(20), SHJ(20),EQC
 SS(20), TB(30), TFK(20), TEA(20), TFC(20),EQC
 TFD(20), TFE(20), THEV(20), XMJAT(20), XNUJ(64),EQC
 COMMON BETA(64,20), EI,J(10,20), GELJ(10,20),EQC

COMMON IC, INFO, INEQV, IP, IRUN, ISC, ISCP1,
 ISMC, ISMCCR, ISR, ISS, ISSNR, ISSP1, ISSP2, ISSP3, ISSP4,
 ISWA, ISWB, ISWPA, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
 ISWB, ISW6A, ISW6B, IUPD, IZERO, JJK, LC, MI, NFI,
 NIT, NNN, NNS, NOS, NQT, NTEST

COMMON IGM(20), ITR(5), KUR(64,20), LPIJ(20,10),EQC
 COMMON ACCM,ELEMENT,HP
 DIMENSION AA(2,24),BTA(64,20),CAPO(31),CCI(20),DGJ(20),SRE
 IJ(20),SDGJ(20),SHJAP(20),THEVP(20)
 EQUIVALENCE (AA(1,1),AAA(1,1)),(BETA(1,1),BT(1,1)),
 (BLBK(1),CAPO(1)),(CAPXTH(1),CCI(1)),(GJA(1),DGJ(1)),
 (CAPX(1),GJ(1)),(SAJ(1),SHJ(1)),(SS(1),SDGJ(1)),
 (SHJA(1),SHJAP(1)),(THEV(1),THEVP(1))

COMMON ZERROR/ERR
 COMMON /EQC/ ZCAP(20),ZSEN,ZCH,ZCM,ZRRP,ZRHO
 COMMON /EQC2/ ZPZ,ZGMU(20)
 DATA RNAVE /6HEQCALC /

CTSAVE=CT
 CT=T/CTAP
 DO 10 I=1,ISC
 ZCAP(I)=QM(I)
 IF (QM(I).LE.0) ZCAP(I)=1.E-3
 ZGMU(I)=ALOG(ZCAP(I))
 ZPZ=ALOG(P)
 CALL THERM
 IF (ISWC.EQ.0) GO TO 60
 DO 30 I=1,ISM
 IL=I+ISC
 CHII(I)=-XMJAT(IL)+ZPZ*BET(I)

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EOC 111
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 EOC 114
 EOC 115
 LQC 116
 EOC 117
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 EOC 119
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 EOC 123
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 EOC 125
 EOC 126
 EOC 127
 EOC 128
 EOC 129-

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ZSEN=-ZPZ
ZCM=0.
DO 230 I=1,ISS
  ZSEN=ZSEN+ZCAP(I)*(SENT(I)-ZGMU(I))
  ZCH=ZCH+ZCAP(I)*SHJ(I)
  ZCM=ZCM+ZCAP(I)*CGI(I)
  ZRHP=P*ZCM/T/R2.055
  ZRHO=ZRBP/(1.+ZRBP*BZERO)
  ZCH=ZCH+(ZRBP/ZRHO-1.)
  CT=CT+SAVE
RETURN (6.250)
CALL DUMP (RNAME)
RETURN
  
```

230
 240
 C
 C
 C
 250

FORMAT (35H100 MANY NEWTON-RAPHSON ITERATIONS)
 END

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SUBROUTINE EXACT
LOGICAL ERR
REAL ACOM(30),ELEMENT(10),HP(20)
DOUBLE PRECISION AA,AAA,CAPX,GJ,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A,AFNTS,AFNX,AMACH,AR,ARBA,ARBB,BZERO,
C,CARB,CH,CHA,CLNT,CM,CMNA,CRA,CTC,CTMX,CTMX,
CRRR,CRS,CSTA,CT,CTAB,CXB,CXMAX,DATEST,OBTEST,DELT1,
CTP,CTPL,CTTI,CT,CTAB,CXB,CXMAX,DATEST,OBTEST,DELT1,
DELT2,DELTA,DLOGA,DLOGR,DLOGR,PRESTH,PRESTH,PRESTH,PRESTH,
SPCTEST,PRES,PRESA,RHOC,RHOP,RHPL,RHPTH,ROBARA,ROBARP,SCPG,SDT,
RHOBAR,RHOC,RHOP,SC,SL,SL64,SU,SU2,SUMG,
SEN,SHPG,SC,SL,SL64,SU,SU2,SUMG,
TEST,TESTB,TPRINT,TSTOP,UP,ZPA
COMMON BE(64),BLBK(31),CGI(20),CGMU(20),CAI(64),CAPXTH(20),
CCPJ(20),CEACT(64),CGI(20),CGMU(20),ETAI(64),ETAJ(20),CHI(64),
CLNIMC(64),CLNPI(64),CMW(20),ETAI(64),ETAJ(20),CHI(64),
GJB(20),PERTGJ(20),PGJ(20),PI(64),PACHI(64),GJAI(20),
QQ(64),SAJ(20),SDCH(64),SENT(20),SHJ(20),SHJA(20),FX,
SKIL(20),SS(20),TFB(30),TFK(20),THEV(20),XNJAT(20),TFEC(20),
TFD(20),TFE(20),TFK(20),THEV(20),XNJAT(20),XNUI(64)
COMMON HETA(64,20),ELJ(10,20),GELJ(10,20)
COMMON XNUIJP(64,20),INEQ,INEQV,IP,IRUN,ISC,ISCP1,
IC,IM,INEQ,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
ISMC,ISMCNR,ISR,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
ISW1A,ISW1B,ISW2A,ISW2B,ISW3A,ISW3B,ISW4A,ISW4B,ISW5A,
ISW5B,ISW6A,ISW6B,IUPD,IZERD,IJJK,LC,MI,NFIT,
NIT,NNN,NNS,NQC,NOT,NTEST
COMMON IGJ(20),IGM(20),ITB(5),KUR(64,20),LPIJ(20,10)
COMMON ACOM,ELEMENT,HP
DIMENSION AAA(22,24),BTA(64,20),CAPO(31),CCI(20),DGJ(20),GJ(20),SBE
IJ(20),SDGJ(20),SHJAP(20),THEVP(20)
EQUIVALENCE (AA(1,1),AAA(1,1)),(BTA(1,1),BTA(1,1)),
(BLXK(1),CAPO(1)),(CAPXTH(1),CCI(1)),(GJA(1),DGJ(1)),
(CAPX(1),GJ(1)),(SAJ(1),SRJ(1)),(SS(1),SDGJ(1)),
(SHJA(1),SHJAP(1)),(THEV(1),THEVP(1))
COMMON /DRV/ SHJDDG
COMMON /ERROR/ ERR
COMMON /SS/ CAS,US
COMMON /TNERK/ SDTE,CTEB,DCHA,CHB,SDCHA,DQMAX,IFAIL
COMMON /TONEQ/ CTE,DTE,BPAR,EPAR(2,25),NT,ITR(25),KTF(25),
KTR(25),ICH(20),IPA(25)
COMMON /TNCE/ SUMGH,SCPGH,GDPR,ODPE
NAMELIST /EXTDMP/ CT,CMA,SU2,DLOGA,DLOGR,SUMG,SHJDDG,SUMTDG,
IUPD,NT,DT,DTE,DGJ,PICHI,SHJ
ZI=SU2/CMA
IF (INEQ.NF.0) GO TO 100
DO 10 I=1,ISSP2
DO 10 J=1,ISSP4
AAA(I,J)=0
DO 20 I=1,ISC
DO 20 J=1,ISS
AAA(I,J)=LPIJ(J,I)
DO 50 I=ISSP1,ISS
K=I-ISC

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 EXA 110

30 S1=BET(K)+SHJ(I)/CT
 DO 30 J=1,ISS
 AAA(I,J)=CDIJ(K,J)*GJ(J)
 S1=S1-CDIJ(K,J)*SHJ(J)/CT
 40 DO 50 J=ISSP1,ISS
 IF (I-J) 50,40,50
 AAA(I,J)=-1.0/GJ(J)
 50 CONTINUE
 AAA(I,ISSP1)=S1/CT
 AAA(I,ISSP2)=RET(K)
 60 CONTINUE
 DO 70 J=1,ISS
 AAA(I,SSPI,J)=1.0
 AAA(I,SSP2,J)=SHJ(J)
 70 AAA(I,SSP1,ISSP1)=SUMG/CT
 AAA(I,SSP1,ISSP2)=SUMG-Z1/CT
 AAA(I,SSP1,ISSP3)=-Z1/CT
 AAA(I,SSP2,ISSP1)=SCPG
 AAA(I,SSP2,ISSP2)=-Z1
 AAA(I,SSP2,ISSP3)=-Z1
 80 DO 80 I=ISSP1,ISSP2
 AAA(I,ISSP3)=-DLOGA*AAA(I,ISSP3)
 CALL DMSOL(AAA,ISSP2,22)
 IF (ERR) RETURN
 DO 90 I=1,ISS
 DGJ(I)=AAA(I,ISSP3)
 90 DT=AAA(I,ISSP1,ISSP3)
 DLOGR=AAA(I,ISSP2,ISSP3)
 GO TO 180
 100 NE=NT+1
 IF (NT.EQ.2) DCHA=-((CMA/CRP/RHAP/US)*QDPR/RHO/SU
 SHJDGJ=0.
 SUMTDG=0.
 DO 120 J=1,ISS
 DGJ(J)=0.
 110 DO 110 I=1,ISR
 DGJ(J)=DGJ(J)+BETA(I,J)*PICHI(I)
 SHJDGJ=SHJDGJ+SHJ(J)*DGJ(J)
 IF (J.LI.NT) GO TO 120
 SUMTDG=SUMTDG+DGJ(J)
 120 CONTINUE
 SUMTDG=CT*SUMTDG
 IF (NT.EQ.2) SUMTDG=SUMTDG+CT*DGJ(1)
 ECOEF=SORT(CMA/8.31434E7/CTAP)/RHOP/SU/CRP
 AAA(1,1)=SCPGH
 IF (NT.EQ.2) AAA(1,3)=CCPJ(1)*GJ(1)
 AAA(1,2)=-Z1
 IF (IUPD.EQ.1) GO TO 130
 AAA(1,NE+1)=Z1*DLOGA-SHJDGJ
 IF (NT.EQ.1) GO TO 130
 AAA(1,4)=AAA(1,4)-ECOE*QDPR
 AAA(2,1)=SUMGH
 IF (NT.FG.2) AAA(2,3)=GJ(1)
 IF (IUPD.EQ.1) GO TO 140
 130 AAA(2,2)=CT*SUMGH-Z1+(NT-1)*GJ(1)*CTE

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140 AAA(2,NE+1)=-SUMTDG+Z1*DLOGA
    IF (NT.EQ.1) GO TO 150
    AAA(3,1)=0.
    AAA(3,3)=1.5*GJ(1)
    IF (IUPD.EQ.1) GO TO 150
    AAA(3,2)=-GJ(1)*CTE
    AAA(3,4)=-1.5*CTE*DGJ(1)+Ecoef*QDPE
150 IF (IUPD.EQ.0) GO TO 170
    AAA(1,NE+1)=Z1*DLOGR-SHJDGJ
    IF (NT.EQ.1) GO TO 160
    AAA(1,4)=AAA(1,4)-Ecoef*QDPR
160 AAA(2,2)=-Z1
    AAA(2,NE+1)=-SUMTDG-DLOGR*(CT*SUMGH-ZI+(NT-1)*GJ(1)*CTE)
    AAA(3,2)=0.
    IF (NT.EQ.1) GO TO 170
    AAA(3,4)=GJ(1)*CTE*DLOGR-1.5*CTE*DGJ(1)+Ecoef*QDPE
170 CALL DSMSOL (AAA,NE,22)
    IF (ERR) RETURN
    DT=AAA(1,NE+1)
    IF (IUPD.EQ.0) DLOGR=AAA(2,NE+1)
    IF (IUPD.EQ.1) DLOGA=AAA(2,NE+1)
    IF (NT.EQ.2) DTE=AAA(3,NE+1)
    IF (DLOGA.NE.0.) GO TO 190
    AMACH=1.
    GO TO 200
190 AMACH=1./SQRT(ABS(1.+DLOGA/DLOGR))
200 IF (INEQ.NE.0.AND.ISW5B.NE.0) WRITE (6,EXTCMP)
    RETURN
    END

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FUNCTION EXP(X)
IMPLICIT REAL*8 (A-H,O-Z)
IF (X.GT.-180.) GO TO 10
EXP=0.D0
GO TO 20
EXP=DEXP(X)
RETURN
END

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140      DADX=DZDX(MBL)
      IF (ABS(1.-AR/A).LE.1.E-5) GO TO 160
      IF (N.LE.10) GO TO 150
      IF (IENTRY.EQ.2.OR..NOT.UPPER) GO TO 150
      V=VL-(VU-VL)*ERRL/(ERRU-ERRL)
      GO TO 60
150      VO=V
      V=V+(A-AR)/ABS(CADX)
      V=AMIN(V,.2.*VO)
      GO TO 60
160      X=SIGN(V,UPDOWN)
      RETURN
C
170      FORMAT (49HCFINDX CALLED WITH AN AREA RATIO LESS THAN UNITY,.10X,2F
180      1HA=,1PE15.8)
      FORMAT (33H0M0PE THAN 50 ITERATIONS IN FINDX,10X,2HA=,1PD15.8,10X,FDX
17HUPDOWN=,0PF3,0,10X,7HENTRY=,12,10X,4HMBL=,12)
      END

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SUBROUTINE FROZEO
REAL ACOM(30),ELEMENT(10),HP(20)
LOGICAL SWITCH,SKIP,MODLPT,ISTEPC,AXISYM,WEDGEM,ERR,LASTPT
DOUBLE PRECISION AA,AAA,CAPX,GJ,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A,AFNLS,AFNX,AMACH,AR,ARBA,ARBB,BZERO,
1  CRB,CRS,CSTA,CLNT,CM, CMA, CRA, CRP,
2  CIP,CTPL,CTT,CT,CTAP,CTB,CTC,CTMAX,CTMXX,
3  DELT2,DELTA,DLOGA,DLOGR,DT,ENT,FLUX,HDELX,DELT1,
5  PCTEST,PRESA,PRESH,PRESTH,PRHQ,RHAP,RHQ,RHOB,
6  RHOBAR,RHOC,RHCP,RHPL,RHTH,ROBARA,ROBARP,SCPG,SDT,
7  SEN,SHPG,SC,SL,SL64,SM,SU,SU2,SUMG,
8  TEST,TESTB,TPRINT,TSTOP,ZP,ZPA
COMMON BE(64),RET(20),BLBK(31),CAI(64),CAPXTH(20),
1  CCPJ(20),CEACT(64),CGI(20),CGMU(20),CHI(64),CHII(20),
2  CLNPI(64),CMW(20),ETAI(64),ETAJ(20),GJA(20),
3  PERTGJ(20),PGI(20),PI(64),PICI(64),OM(20),
4  SAJ(20),SDCHI(64),SENT(20),SHJ(20),SHJA(20),
5  SS(20),TFE(20),TFB(30),TFA(20),TFC(20),
6  TFE(20),TFK(20),THEV(20),THEV(20),XNJAT(20),
COMMON BETA(64,20),ELJ(10,20),GELJ(10,20),
1  XNUIJ(64,20),XNUIJP(64,20)
COMMON IC,INEQ,INEOV,IP,IRUN,ISC,ISCP1,
1  ISAC,ISMCNR,ISR,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
2  ISW1A,ISW1B,ISW2A,ISW2B,ISW3A,ISW3R,ISW4A,ISW4B,ISW5A,
3  ISW5B,ISW6A,ISW6B,IUPD,IZERO,JJK,LC,MI,NFIT,
4  NIT,NNN,NNS,NQS,NQT,NTEST
COMMON IGJ(20),IGM(20),ITB(5),KUR(64,20),LPIJ(20,10)
COMMON ACOM,ELEMENT,HP
COMMON AA(22,24),BTAA(64,20),CAPO(31),CCI(20),DGJ(20),GJ(20),SBEFRO
1  J(20),SDGJ(20),SHJAP(20),THEVP(20)
EQUIVALENCE (AA(1,1),AAA(1,1)),(BETA(1,1),BTA(1,1)),
1  (BLBK(1),CAPO(1)),(CAPXTH(1),CCI(1)),(GJA(1),DGJ(1)),
2  (CAPX(1),GJ(1)),(SAJ(1),SBJ(1)),(SS(1),SDGJ(1)),
3  (SHJA(1),SHJAP(1)),(THEV(1),THEVP(1))
COMMON /TEMPRY/SAVEC(20)
COMMON /AREA/ATPI(1,2),PARAM(3,12,2),RTHCM(2),NSECT(2),
1  NSFCTU(2),ISHAPE(12,2),NPROFL(2),NPFPLS,NBL
COMMON /BLOUT/REPF,THETA(2),SN(2),XSN(2),PRREF,HR,QWDOT(2),
1  TAUW(2)
COMMON /RESRPP/ VISC,PRR,SIGR,FLEWR
COMMON /TRRPP/ VISC,PR,SIGMA,FLEWIS
COMMON /ERROR/ ERH
DIMENSION DELBLO(2),DELBLP(2)
COMMON /MODPAR/ XMP1,DXMP,FSTAG, CATFAC,TMODEL,XMODP1,DXMGDP,
1  TPLATE,KDIN
COMMON /STAG/ T2E,EPSLCN,T2F,FPSE
COMMON /MCDPT/ TSDIAM(20),TSAR(20),NTS,MBL
DIMENSION BLVAR(13),BLVARC(13),BLVAR(13)
EQUIVALENCE (BLVAR(1),REPF)
DIMENSION SOLN(3,2)
DATA SOLN /4HFROZ,4HEN F,3HLOW,4HEQUI,4HLIBR,3HIUM/
DATA ONE /1./

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COMMON /BL/ DELBL(2),BLINT(2),XZERO,CPWALL,VISROT,DIAM(2),
SW,RO,JDIM,IPCINT
COMMON /SS/ CAS,US
COMMON /NEO/ CMDST(2),ODELBL(2)
COMMON /SW.TCH/ XPB,DXPB,W
COMMON /AVG/ WSAVE
COMMON /RDWEDG/ ANGLE(10),PADLE(5),WX1,DWX,WXI(20),TWEDGE,WK,
NWX,NANGLE,NRAOLE,WEDGEM,AXISYM,ISW9B
1 NAMELIST /FRDMP/ CT,ISONIC,FLAG,PRES,RHO,SU,AMACH,AFNX,XLAST,
DELBL,DEBLO,XO,DELBLP,XP,BLVAR,IRSTR,W
1 NAMELIST /FRDYP2/ XMODEL,X1,X2,CT,CT1,CT2,NMPI,XFRAC,DELBL,
DEBLO,ERRX
C FROZEN FLOW SOLUTION
ENTRY FROZEN
ISOLN=1
DETERMINE SONIC CONDITIONS IN FROZEN FLOW
DELT=DELT1
FI=0.0
TI=1.0
CT=TI-DELT
SI=0.0
CLNMA=ALOG(CMA)
DO IC J=1,ISS
SS(J)=ALOG(GJA(J))
SI=SI-GJA(J)*(SS(J)+CLNMA)
S2=-(ZPA/CMA)
ENT=SI+S2-CRRB
T2=CT
CALL THERM
CALL PROP
F2=FLUX
CT=CT-DELT
T3=CT
CALL THERM
CALL PROP
F3=FLUX
IF (F3-F2) 40,60,80
IF (DELT-TESTB) 90,50,50
DELT=DELT/2.0
CT=TI-DELT
GO TO 20
IF (DELT-TESTB) 90,70,70
TI=T2
FI=F2
DELT=DELT/2.0
CT=TI-DELT
GO TO 2C
TI=T2
T2=T3
FI=F2
F2=F3
GO TO 30
Z=F1-F2+F3-F2
IF (AHS(Z/F2).GT..00001) GO TO 100
CT=T2

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100 GO TO 110
110 CT=I2+DELT*(F3-F1)/(2.0*Z)
111 CONTINUE
112 CALL THERM
113 CALL PRCP
114 S=FLUX
115 CTMAX=CT
116 FRESTH=PRES
117 RHTH=PHC
118 IPASS=2
119 GO TO 120
120 EQUILIBRIUM FLOW SOLUTION
121 ENTRY EQUIL(IPASS)
122 ISCLN=2
123 IRSIRT=1
124 W=SAVE
125 DO 140 J=1,ISS
126 GJ(J)=GJA(J)
127 ISONIC=-1
128 LASTPT=.FALSE.
129 SKIP=.FALSE.
130 DELTSV=DELT1
131 IF (ISW3H.NE.0) DELT1=0.049*(1.-CTMAX)
132 XMCDI=XMPI
133 ITSE=1
134 ITSTEP=.FALSE.
135 MODLPT=.FALSE.
136 CALL NEXTMP (ITS,XMCDI,XMODEL)
137 INECC
138 T2F=CTAP
139 DO 150 I=1,13
140 BLVAR(I)=0.
141 EPSLON=C.
142 T2F=CTAP
143 EPSFEU.
144 DO 160 I=1,NPRFLS
145 CCLBL(I)=0.
146 OMDST(I)=1.
147 ISWICH=0
148 SWTCH=.FALSE.
149 IPOINT=1
150 PRES=1.C
151 CT=1.0
152 RHU=1.0
153 RHOCAP=ROBARA
154 SU=0.0
155 AVACH=0.0
156 AFNXC=0.0
157 RHOB=1.0
158 PRESB=1.0
159 DELTV=DELT1
160 XO=0.
161 XP=0.
162 FLAG=-1.
163 IF (IPASS.EQ.1) GO TO 180

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170 WRITE (6,630) (ISCLN(N,ISCLN),N=1,3)
DO 170 J=1,ISS
SAVEC(J)=GJA(J)*CMA
AMACH=0.
AFNX=1.F38
VICC=VISC
SIGMA=SIGR
CX=-1.E10
CM=CHA
CM=CMA
CALL OUT2 (ISCLN)
IF (ERR) RETURN
DECREMENT TEMPERATURE TO GENERATE SOLUTION
IF (.NOT. TSTEPC.AND.(XO-XP).LT.10.) GO TO 190
TSTEPC=.TRUE.
DELTA=AMINI*(DELTA1,10.*DELTA/(XO-XP))
DELTA=AMAXI*(DELTA,C.02*DELTA1)
DELTA=AMINI*(DELTA,0.1*CT)
CT=CT-DELTA
IF (ISONIC.EQ.-1.AND.CT.LE.CTMAX) GO TO 200
IF (CT.LT.YSTOP) GO TO 610
GO TO 220
THROAT HAS BEEN REACHED. COMPUTE CONDITIONS AT THROAT.
IF (ISW3B.NF.0) GO TO 210
FLAG=1.
ISONIC=1
GO TO 220
CT=STARCT+DELTA
CT=CTMAX
ISONIC=0
COMPUTE THE FLOW CONDITIONS FOR TEMPERATURE CT.
GO TO (230,240), ISCLN
CALL THERM
CALL PROC
GO TO 250
CALL NEWRAP
IF (ERR) RETURN
AFNX=AMAXI(SM/FLUX,ONE)
IF (PRES.EQ.PRES) GO TO 260
AMACH=SU*SQRT(RHO*ABS(ALCG(RHO/RHOB)/ALCG(PRES/PRESB)))/PRES)
PRES=PRES
SM=SM/RHO
ASE=SU*US/AMACH
CASEAS=AS/CT
DO 290 J=1,ISS
GO TO (270,280), ISCLN
SAVEC(J)=GJA(J)*CMA
GO TO 290
SAVEC(J)=GJ(J)*CM
CONTINUE
IF (SKIP) GO TO 360
DETERMINE X FROM EFFECTIVE AREA RATIO
IF (ISW3B.EQ.0) GO TO 310
IF (ISONIC) 310,320,300
IF (SWTCH) GO TO 340

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310 IS .CH=ISWTC*1
    IF (ISWTC.EQ.3) GO TO 330
    X BASED ON APPROXIMATING GEOMETRIC AREA RATIO BY AFNX
    CALL FINDX (AFNX,FLAG,CX)
    IF (EKRR) RETURN
    IF (ISW3B.EQ.0) GO TO 410
    IF (CX.LT.XZERO) GO TO 180
    GO TO 370
320 CX=0.
    DELTI=DELTSV
    DELTV=DELT1
    GO TO 370
330 SWTC=.TRUE.
    XPB=XO
    DXPB=XO-XP
    X CORRECTED FOR DISPLACEMENT THICKNESS
    CALL AGSOLN (AFNX,DELBL,FLAG,AR,CX)
    IF (ERR) RETURN
    IF (MODLPT) GO TO 360
    IF (CX.GT.XLAST) GO TO 370
    WRITE (6,640)
    IF (ISW3B.EQ.0.OR.AR.GT.4.*AFNX) GO TO 350
    IRSTRT=IRSTRT+1
    IF (IRSTRT.GT.3) GO TO 350
    W=W/2.
    WRITE (6,650) W
    GO TO 130
    WRITE (6,FRCMP)
    RETURN
350 CALL TRANSP (CT*CTAP,PRES*PRESA)
    GO TO 410
    XLAST=CX
    CALL BLAYER (.TRUE.)
    IF (ERR) RETURN
    IF (IPASS.EQ.1.AND.ISONIC.NE.0) GO TO 180
    IF (ISW3B.EQ.0) GO TO 400
    DO 390 I=1,NPRFLS
    QMDS(I)=DELBI(M)
    IF (IPASS.EQ.1) RETURN
    CALL AGSOLN (AFNX,DELBL,FLAG,AR,CX)
    IF (MODLPT) GO TO 500
    IF (.NOT.SKIP) GO TO 420
    SKIP=.FALSE.
    GO TO 550
    XP=XO
    XD=CX
    IF (ISW3B.EQ.0) GO TO 440
    DO 430 M=1,NPRFLS
    DELBLP(M)=DELBLC(M)
    DELRLO(M)=DELBL(M)
    IF (CX.LT.XMAX) GO TO 450
    XMODEL=CX*CX
    LASTPT=.TRUE.
    IF (CX.GE.XMODEL) GO TO 460
    IF (SKIP) GO TO 220
  
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460 GO TO 550
    X1=XP
    X2=XD
    MODLPT=.TRUE.
    SKIP=.FALSE.
    NMPI=0
    CT1=CT+DELT
    CT2=CT
    CTSAVE=CT
    IF (ISWB.EQ.0) GO TO 490
    CXSAVE=CX
    XFRAC=(XMODEL-XP)/(XO-XP)
    DO 470 M=1,NPRFLS
    DELBL(M)=(DELBLC(M)-DELBLP(M))*XFRAC+DELBLP(M)
470 DO 480 K=1,13
    BLVARS(K)=BLVAR(K)
480 CT=CT1+(CT2-CT1)*(XMODEL-X1)/(X2-X1)
490 GO TO 220
    NMPI=NMPI+1
    ERRX=CX-XMODEL
    IF (ABS(ERRX).LE.0.0025) GO TO 520
    IF (NMPI.LE.9) GO TO 510
    WRITE (6,660) CX,XMODEL
    GO TO 520
510 CT1=CT2
    CT2=CT
    X1=X2
    X2=CX
    GO TO 490
520 IF (LASTPT) GO TO 530
    CALL NEXTMP (ITS,XMOD1,XMODEL)
530 SKIP=.TRUE.
    IF (ISWB.EQ.0) GO TO 550
    DO 540 K=1,13
    BLVAR(K)=BLVARO(K)+(BLVARS(K)-BLVARO(K))*XFRAC
540 CALL OUT2 (ISOLN)
550 IF (ERR) RETURN
    IF (.NOT.MODLPT) GO TO 580
    MODLPT=.FALSE.
    CALL MDEL
    IF (LASTPT) GO TO 610
    IF (ERR) RETURN
    CT=CTSAVE
    CX=CXSAVE
    IF (ISWB.EQ.0) GO TO 440
560 DO 560 K=1,13
    BLVAR(K)=BLVARS(K)
570 DO 570 M=1,NPRFLS
    DELBL(M)=DELBLP(M)
    GO TO 440
580 IF (ISWB.EQ.0) GO TO 600
    DO 590 K=1,13
590 BLVARO(K)=BLVAR(K)
600 IF (CX.GT.CXMAX) GO TO 610
  
```

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IF (ISONIC,NE,0) GO TO 180
CT=CTSTAR
ISONIC=1
FLAG=1,
GO TO 180
IF (NPRFLS,EO,1) GO TO 620
AXISYM=.TRUE.
CALL MDEL
AXISYM=.FALSE.
RETURN

        FORMAT (1H1,3A4,8HSOLUTION//)
        FORMAT (2PHOX DECREASED IN FROZEG)
        FORMAT (103H RESTART SOLUTION WITH A LARGER AVERAGING DISTANCE FORFRO
1 THE BOUNDARY LAYER CORRELATION PARAMETER. NEW W=.F9.6/1H1)
        FORMAT (73H ITERATION TO OBTAIN FREE-STREAM SOLUTION AT MODEL POINTFRO
IT DID NOT CONVERGE,5X,2HX=.1PE10.3,5X,7HXMODEL=.E10.3,3H CM)
END

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SUBROUTINE GEOM
REAL ACCM(30),ELMENT(10),HP(20)
LOGICAL ERR
DOUBLE PRECISION AA,CAPX,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A, AFNTS, AFNX, AMACH,
C, CARB, CRB, CSTA, CT, CTAP, CLNT,
1 CRRB, CRS, CTT, CX, CXB, CXMAX,DATEST,DBTEST,DELT1,
2 CTP, CTPL, DLOGA, DLOGR, DT, ENT, FLUX, HDELX, PCT,
3 DELT2,DELTA, DLOGA, DLOGR, DT, ENT, FLUX, HDELX, PCT,
5 PCTEST, PRES, PRESA, RHOC, RHOP, RHOB, RHOC, RHOD, RHOB,
6 RHOC, RHOD, RHOB, RHOC, RHOD, RHOB, RHOC, RHOD,
7 SEN, SHPG, SC, SL, SL64, SM, SU, SU2,
8 TEST, TFSTB, TPRINT, TSTOP, UP, ZPA
COMMON RE(64), BLAK(31), CAI(64),CAPXTH(20),GEO
1 CCPJ(20), CEACT(64), CGI(20), CGMU(20), CHI(64), CHI(20),GEO
2C1 NIMC(64), CLNEI(64), CMW(20), ETAI(64), ETAJ(20),GEO
3 GJB(20),PERTGJ(20), PGJ(20), PI(64), PICH(64), QM(20),GEO
4 GQ(64), SAJ(20), SDCHI(64), SENT(20), SHJ(20), SHJA(20),GEO
5 SKIL(20), SS(20), TB(30), TFA(20), TFH(20), TFC(20),GEO
6 TFD(20), TFE(20), TFK(20), THEV(20), XMJAT(20), XNUI(64)
COMMON BETA(64,20), FLJ(10,20), GELJ(10,20),GEO
1 XNUIJ(64,20),XNUIJP(64,20)
COMMON IC, IM, INEQ, INEQV, IP, IRUN, ISC, ISCP1,
1 ISMC, ISMCNR, ISR, ISS, ISSNR, ISSP1, ISSP2, ISSP3, ISSP4,
2 ISW1A, ISW1B, ISW2A, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
3 ISW5B, ISW6A, ISW6B, IUPD, IZERO, IZK, LC, MI, NFI,
4 NIT, NNS, NOS, NOT, NTEST
COMMON IGJ(20), IGM(20), ITR(5), KUR(64,20), LPIJ(20,10)
COMMON ACOM,ELMENT,HP
COMMON /BL/ DELBL(2),BLINT(2),XZERO,TWALL,CPWALL,VISROT,DIAM(2),
1 SA,RO,JOIM,IPCINT
COMMON /NEO/ OMCST(2),DDELBL(2)
COMMON /AREA/ ATPI(11,2),PARAM(3,12,2),RTHCM(2),NSECT(2),
1 NSECTU(2),ISHAPE(12,2),NPROFL(2),NPRFLS,NBL
COMMON /AEGEOM/ SORTA,S1,S2
COMMON /ERROR/ ERR
DIMENSION DELSTP(2)
DATA RNAME /4HGFEOM/
CALL GEDMAR (CX,S1,S2)
IF (ERR) RETURN
IF (NPRFLS.EQ.1.AND.JDIM.EQ.1) SORTA=SQRT(S1)
IF (ISW3B.EQ.0.CR,IUPD.EQ.1) GO TO 10
CALL AESCLN (CX)
IF (INEQ.EQ.0.OR,IUPD.EQ.0) GO TO 180
IF (S1.LE.1.D0) GO TO 50
ARG=2.*C*(S1-1.D0)/A
DEL RHO=SQRT(ARG)
IF (S1.GT.1.001) GO TO 60
IF (S2) 20,20,30
SGN=1.
GO TO 40
SGN=-1.
30
RHO=RHOTH+SGN*SQRT(ARG-SGN 1.)/J.*SQRT(A/C/(A+2.))*DEL RHO*#3)
GO TO 130

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S1=1.
S2=C.
RHO=RHTH
GO TO 130
A2=S1*S1
N=0
OPHA=1.+0.5*A
IF (S2) 70,70,80
RHO=A*MIN1(RHTH+DELRHO,(1.-C/A2)**(1./A))
GO TO 90
RHO=A*MAX1(RHTH-DELRHO, SORT(C)/S1)
N=N+1
IF (N-50) 100,100,130
CONTINUE
IF (RHO.GE.0.) GO TO 110
WRITE (6,200)
CALL DUMP (RNAME)
RETURN
Z=0. **A
OMZ=1.-Z
AN=SQRT(C/OMZ)/RHO
DADR=AN/RHO*(OPHA#Z-1.)/OMZ
DD=(S1-AN)/DADR
RHO=RHO+DD
IF (ABS(DD/RHO).GE.1.0E-6) GO TO 90
Y=SIGN(1.0DD,CX)*(RHO-RHTH)
IF (Y) 130,130,120
RHO=RHTH-SIGN(1.0D-02,CX)
N=0
GO TO 90
AFNX=SM/(SU*RHQ)
IF (S2.NE.0.) GO TO 170
IF (NPRFLS.EQ.1) GO TO 150
D2LA=0.
DO 140 J=1,2
DELSTP(J)=RO*(1.-OMDST(J))
NSUJ=NSECTU(J)
D2LA=D2LA+1./((RTHCM(J)-DELSTP(J))/PARAM(3,NSUJ,J))
GO TO 160
NSUJ=NSECTU(1)
D2LA=1./((RTHCM(1)*OMDST(1))*PARAM(3,NSUJ,1))
IF (JDIM.NE.0) D2LA=2.*D2LA/SORTA
DLOGR=-SQRT(C*D2LA/A/RHTH)
GO TO 190
DLOGR=((12.*C)/(A*(S1*RHQ)**2-C*(A+2.)))*S2/S1
GO TO 190
AFNX=S1
DLOGA=S2/S1
RHQ=SM/(SU*AFNX)
RETURN
FORMAT (21HNEGATIVE RHO IN GEOM)
END

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SUBROUTINE GEOMAR (X,ARATIO,DERIVA)
COMPUTES GEOMETRIC AREA RATIO ARATIO AND ITS DERIVATIVE DERIVA.
REAL ACOM(30),CLMENT(10),HP(20)
DOUBLE PRECISION AA,CAPX,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A, AFNTS, AFNX, AMACH, AR, ARBA, ARBB, BZERO,
C, CARB, CH, CHA, CTAP, CTAT, CTB, CMA, CRA, CRP,
1 CRRB, CRS, CSTA, CTI, CTX, CXC, CXMAX, DATEST, DBTEST, DELTI,
2 CTP, CTPL, CTT, CX, DT, ENT, FLUX, HDELX, PCT,
3 DELT2, DELTAX, DLOGA, DLOGR, PRESA, PRESB, PRESTH, PRHO, RHAP, RHO,
4 SPECTEST, RHOC, RHOP, RHPL, RHTH, ROBAR, ROBARP, SCPG, SDT,
5 GRHOBAR, RHOC, RHOP, RHPL, RHTH, ROBAR, ROBARP, SCPG, SDT,
6 SEN, SHPG, SC, SL, SL54, SM, SU, SU2, SUMG,
7 TEST, TESTB, TPRINT, TSTOP, ZP, ZPA
8 COMMON BE(64), RET(20), RLK(31), CAI(64), CAPXTH(20),
1 CCPJ(20), CFACT(64), CGI(20), CGMU(20), CHI(64),
2 CLN1, MC(64), CLNFI(64), CM*(20), ETAI(64), ETAJ(20),
3 GJB(20), PERTGJ(20), PGJ(20), PI(64), PCHI(64),
4 QU(64), SAJ(20), SDCHI(64), SFNT(20), SHJ(20),
5 SKIL(20), SS(20), TFA(20), TFB(30), TFC(20),
6 TFD(20), TFE(20), TFK(20), THEV(20), XMJAT(20),
COMMON BETA(64,20), ELJ(10,20), GELJ(10,20),
1 XNUIJ(64,20), XNUIJP(64,20)
COMMON IC, IM, INEQ, INEQV, IP, IRUN, ISC, ISCP1,
1 ISMC, ISMCNR, ISR, ISS, ISSNR, ISSP1, ISSP2, ISSP3, ISSP4,
2 ISW1A, ISW1R, ISW2A, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
3 ISW5R, ISW6A, ISW6B, IURD, IZERO, JJK, LC, M1, NFIT,
4 NIT, NNN, NNS, NOS, NQT, NTEST
COMMON IGM(20), ITR(5), KUR(64,20), LPIJ(20,10)
COMMON ACOM, ELEMENT, HP
COMMON /AREA/ ATPI(11,2), PARAM(3,12,2), RTHCM(2), NSECT(2),
1 NSECTU(2), ISHAPE(12,2), NPROFL(2), NPRFLS, NBL
COMMON /BL/ DELFL(2), BLINT(2), XZERO, TWALL, CPWALL, VISROT, DIAM(2),
1 SW, RC, JDIN, IPCINT
DIMENSION SN(2), RAD(2), S1(2), S2(2), DRV(2)
DATA HNAME /'GHGEOGAR/'
DATA SN /-1.,1./
COMMON /THRT/ PSA
ENTRY=1
GO TO 10
ENTRY GMAR(X,Y)
ENTRY=2
GO TO 10
ENTRY GMAR2(X,Y,Z)
ENTRY=3
GO TO 10
ENTRY GMAR3(X,OYDX,OZDX,Y,Z)
ENTRY=4
DO 100 I=1,NPRFLS
NSM1=NSECT(I)-1
DO 20 J=1,NSM1
K=J
IF (X,LT,ATPI(J,1)) GO TO 30
CONTINUE
K=NSECT(I)

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7  ISJ=ISHAPE(K,I)
   IF (ISJ.GT.1) GO TO 7C
   RAD(I)=PARAM(1,K,I)+PARAM(2,K,I)*X
   GO TO (50,100,100,40), IENTRY
40  DRV(I)=PARAM(2,K,I)
   GO TO 100
50  RATERAD(I)/RTHCM(I)
   IF (JDIM.EQ.0.OR.NPRFLS.EQ.2) GO TO 60
   S1(I)=RAT*#2
   S2(I)=2.*RAT/RTHCM(I)*PARAM(2,K,I)
   GO TO 140
60  S1(I)=RAT
   S2(I)=PARAM(2,K,I)/RTHCM(I)
   GO TO 100
70  DAB=X-PARAM(2,K,I)
   DES RT(PARAM(3,K,I))*#2-DAB**2)
   RAD(I)=PARAM(1,K,I)+SN( ISJ-1)*Q
   GO TO (80,100,100,80), IENTRY
80  DRV(I)=-SN( ISJ-1)*DAB/Q
   IF (IENTRY.EQ.4) GO TO 100
   RATERAC(I)/PTHCM(I)
   IF (JDIM.EQ.0.OR.NPRFLS.EQ.2) GO TO 90
   S1(I)=RAT*#2
   S2(I)=2.*RAT/RTHCM(I)*DRV(I)
   GO TO 100
90  S1(I)=RAT
   S2(I)=DRV(I)/RTHCM(I)
100  CONTINUE
   IF (NPRFLS.EQ.1) GO TO 130
   GO TO (120,130,110,110), IENTRY
110  SOTRSA=SQRT(RSA)
   Y=RAD(1)*SOTRSA
   Z=RAD(2)*SOTRSA
   IF (IENTRY.EQ.2) GO TO 180
   DYDX=DRV(1)*SOTRSA
   DZDX=DRV(2)*SOTRSA
   GO TO 180
120  ARATIO=S1(1)*S1(2)*RSA
   DERIVA=(S1(1)*S2(2)+S1(2)*S2(1))*RSA
   GO TO 150
130  IF (IENTRY.EQ.2) GO TO 160
140  ARATIO=S1(1)*RSA
   DERIVA=S2(1)*RSA
150  IF (ARATIO.LT.1.,.AND.ARATIO.GE.0.999999) ARATIO=1.
   IF (ARATIO.GE.1.,.AND.X*DERIVA.GE.0.) GO TO 180
   WRITE (6,190) X,ARATIO,DERIVA
   CALL DUMP (XNAME)
   RETURN
160  Y=RAD(1)
   IF (JDIM.EQ.0) GO TO 170
   Y=Y*SQRT(RSA)
   GO TO 180
170  Y=Y*RSA
180  RETURN
C
  
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GMR 111
GMR 112
GMR 113
GMR 114
GMR 115-

FORMAT (40HCERROR IN INPUT DATA FOR NOZZLE GEOMETRY.4H X=.1PE14.6
1.9H ARATIO=.E14.6.9H DERIVA=.E14.6)
END

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190


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56 BET(I)=-1.0
57 DO 40 K=1,ISC
58 BET(I)=RET(I)+CDIJ(I,K)
59 CONTINUE
60 SUM=0.0
61 DO 80 I=1,ISC
62 OM(I)=0.0
63 DO 70 K=1,ISC
64 OM(I)=OM(I)+AIN(K,I)*CAPQ(K)
65 SUM=SUM+OM(I)
66 CONTINUE
67 DO 90 I=1,ISC
68 OM(I)=OM(I)/SUM
69 CCI(I)=0.0
70 DO 90 J=1,ISC
71 CCI(I)=CCI(I)+LPIJ(I,J)*CMW(J)
72 DO 100 I=1,ISC
73 CGI(I)=CCI(I)
74 IF (ISMC.EQ.0) GO TO 130
75 DO 120 I=1,ISMC
76 SUM=0.0
77 DO 110 K=1,ISC
78 SUM=SUM+CDIJ(I,K)*CGI(K)
79 IL=I+ISC
80 CGI(IL)=SUM
81 CONTINUE
82 RETURN
83 ENTRY INTA
84 CALL EQCALC (CTAP,PRESA)
85 IF (ERR) RETURN
86 ZPA=ZPZ
87 CM=ZCM
88 SEN=ZSEN
89 CHA=ZCH
90 CMA=ZCM
91 CRR=SEN/CMA
92 SEN=CRR*#CRA
93 CHECHA
94 DO 140 I=1,ISS
95 CGMU(I)=ZGMU(I)
96 CAPX(I)=ZCAP(I)
97 GJA(I)=CAPX(I)/CMA
98 ROBARP=ZRHYP
99 RHAP=ZRHG
100 ROBARA=ROPARP/RHAP
101 RETURN
102 END

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SUBROUTINE INIT
REAL ACOX(30),ELEMENT(10),HP(20)
DOUBLE PRECISION AA,AAA,CAPX,GJ,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A, AFNTS, CHA, CLNT, AMACH,
1 C, CARB, CH, AR, ARBA, ARBB, BZERO,
2 CRB, CRS, CSTA, CT, CTAP, CM, CMA, CRA, CRP,
3 CTP, CTPL, CTT, CX, CXB, CTB, CTC, CTMAX, CTMXX,
4 DELT2, DELTAX, DLOGA, OLOGA, OT, ENT, FLUX, HOELX, DELTI,
5 PCTEST, PRES, PRESA, RHOC, RHOP, RHOC, RHAP, RHO, RHOB,
6 PHOBAR, RHOC, RHOP, RHOC, RHAP, RHO, RHOB,
7 SEN, SHPG, SHOC, SC, SL, SL64, SM, SDT,
8 TEST, TESTB, TPRINT, TSTOP, UP, ZPA, SUMG,
COMMON RE(64), RET(20), BLBK(31), CAI(64), CAPXTH(20),
1 CCPJ(20), CEACT(64), CMI(20), CGMU(20), CHI(64), CHI(20),
2 CLN1VC(64), CLNPI(64), CGW(20), ETAI(64), FTAJ(20), GJA(20),
3 GJT(20),PERTGJ(20), PGJ(20), PI(64), PICH(64), QM(20),
4 GJ(64), SAJ(20), SDCHI(64), SENT(20), SHJ(20), SHJA(20),
5 SKIL(20), SS(20), TH(30), TFA(20), TFB(20), SHJA(20),
6 TFE(20), TFK(20), THEV(20), XMJAT(20), TFC(20),
COMMON RETA(64,20), ELJ(10,20), GELJ(10,20), XNUI(64)
1 XNUIJ(64,20),XNUIJP(64,20)
COMMON IC, IM, INEQ, IP, IRUN, ISC, ISCP1,
1 ISMC, ISV, ISR, ISS, ISSNP, ISSP2, ISSP3, ISSP4,
2 ISK1A, ISW1B, ISW2A, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
3 ISW5B, ISW6A, ISW6B, IUPD, IZRO, JJK, LC, MI, NFIT,
4 NIT, NNN, NNS, NOS, NQT, NTEST
COMMON IGJ(20), IGM(20), ITR(5), KUR(64,20), LPIJ(20,10)
COMMON ACOV,ELEMENT,HP
DIMENSION AAA(22,24),RTA(64,20),CAPQ(31),CCI(20),DGJ(20),GJ(20),SB
1J(20),SDGJ(20),SHJAP(20),THEVP(20)
EQUIVALENCE (AAA(1,1),AAA(1,1)),(BETA(1,1),BETA(1,1)),
1 (BLBK(1),CAPQ(1)),(CAPXTH(1),CCI(1)),(GJA(1),DGJ(1)),
2 (CAPX(1),GJ(1)),(SAJ(1),SHJ(1)),(SS(1),SDGJ(1)),
3 (SHJA(1),SHJAP(1)),(THEV(1),THEVP(1))
COMMON /WASSFL/ SMASS,CTMXXI,TSTOPI,IS(20)
DIMENSION SPRB(43,30)
COMMON /POLYAT/ THLVE(4,20)
COMMON /SPEC/ SPI(43),SP2(43),SP3(43),SP4(43),SP5(43),SP6(43),
1 SP7(43),SP8(43),SP9(43),SP10(43),SP11(43),SP12(43),SP13(43),
2 SP14(43),SP15(43),SP16(43),SP17(43),SP18(43),SP19(43),
3 SP20(43),SP21(43),SP22(43),SP23(43),SP24(43),SP25(43),
4 SP26(43),SP27(43),SP28(43),SP29(43),SP30(43)
EQUIVALENCE (SPI(1),SPRP(1,1))
COMMON /TNONEQ/ CTE,DTE,BPAR,EPAR(2,25),NT,ITR(25),KTF(25),
1 KTR(25),ICH(20),IPA(25)
CTMXX=CTMXXI/CTAP
IF (INEOV,NE,C) CTMXX=2
TSTOPI=TSTOPI/CTAP
IF (TSTOP,EQ,0.) TSTOP=0.004
ARRA=0
ARRB=0
DATEST=1.01
OBTST=5.0
PCTEST=1.2

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SUBROUTINE KANDMU (ZK)
SUBROUTINE KANDMU COMPUTES VISCOSITY AND TRANSLATIONAL COMPONENT
OF THERMAL CONDUCTIVITY. INDEX L=1 INDICATES TRANSLATIONAL
THERMAL CONDUCTIVITY, L=2 INDICATES VISCOSITY.
DIMENSION ZK(2)
COMMON /TRANS1/ T,Q(3,20,20),ZM2(20)
COMMON /TRANS3/ B(20,2),BR(20,2),A(2),X(20),DH(20)
COMMON /TRANS5/ N,IELEC,IDI,ID2
COMPUTE DIAGONAL MATRIX ELEMENTS CAP-A(I)
DO 20 I=1,N
  R(I,1)=C.
DO 10 J=1,N
  B(I,1)=R(I,1)+X(J)*Q(3,I,J)
  B(I,2)=B(I,2)/ZM2(I)
BR(I,1)=1./B(I,1)
BR(I,2)=1./B(I,2)
IF (N.EQ.1) GO TO 70
COMPUTE AVERAGE OF NON-DIAGONAL MATRIX ELEMENTS SMALL-A(I,J)
DO 60 L=1,2
  Y=0.
DO 40 I=2,N
  II=I-1
  V=0.
  W=0.
DO 30 J=1,II
  REX(J)=(BR(I,L)-BR(J,L))*#2
  V=V+R
  W=W+R*Q(L,I,J)
  U=U+X(I)*W
  Y=Y+X(I)*V
IF (Y.EQ.0.) GO TO 50
A(I)=U/Y
GO TO 50
A(L)=C.
CONTINUE
A(1)=5*A(1)
COMPUTE VISCOSITY AND TRANSLATIONAL THERMAL CONDUCTIVITY
DO 90 L=1,2
  Y=0.
DO 80 I=1,N
  Y=Y+X(I)/(A(L)+B(I,L))
  ZK(L)=3.125*Y/(1.-A(L))*Y
RETURN
END

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SUBROUTINE KINT (ZKINT)
COMMON /TRANS1/ T,Q(3,20,20),ZM2(20)
COMMON /TRANS3/ B(20,2),BR(20,2),A(2),X(20),DH(20)
COMMON /TRANS5/ N,IELEC,IO1,IO2
COMPUTE SUM OVER J OF X(J)*DELTA(I,J)
IF (N.EQ.1) GO TO 20
N1=N+1
DO 10 I=2,N
  I1=I-1
  N1=N-I1
  DO 10 J=1,I1
    NJ=N1-J
    DO 10 L=1,2
      B(I,L)=B(I,L)+X(J)*Q(L,J,I)
      B(N1,L)=B(N1,L)+X(NJ)*Q(L,N1,NJ)
COMPUTE THERMAL CONDUCTIVITY COMPONENT DUE TO INTERNAL EXCITATION
ENERGY
ZKINT=0.
DO 30 I=1,N
  ZKINT=ZKINT+DH(I)*X(I)/B(I,I)
RETURN
END

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SUBROUTINE LIST
LOGICAL SUP30,WEDGEM,AXISYM
REAL A(20),ELVNT(10),HP(20)
DOUBLE PRECISION AA,AAA,CAPX,GJ,COIJ
COMMON AA(2,24),CDIJ(20,10),CAPX(20)
COMMON C, CARB, CSTA, CH, ARBA, ARBB, BZERO,
: 2 CRRB, CRS, CTT, CX, CTB, CTM, CTMA, CTMAX, DATEST, DBTEST, DELT1,
3 CTP, CTPL, CTT, CX, CTB, CTM, CTMA, CTMAX, DATEST, DBTEST, DELT1,
4 DELT2, DELTAX, DLOGA, DLOGR, PRESB, PRESTH, RHPL, RHTH, ROBARA, ROBARP, SCRG, SDT,
5 SPCTEST, PCC, RHOP, RHPL, RHTH, ROBARA, ROBARP, SCRG, SDT,
6 RTOBAR, RTOCC, SHPG, SC, TSTP, TSTPB, TPRINT, TSTP, TSTPB, TPRINT, TSTP, TSTPB, TPRINT,
7 SEN, SHPG, SC, TSTP, TSTPB, TPRINT, TSTP, TSTPB, TPRINT, TSTP, TSTPB, TPRINT,
8 TEST, TESTB, TPRINT, TSTP, TSTPB, TPRINT, TSTP, TSTPB, TPRINT, TSTP, TSTPB, TPRINT,
COMMON BE(64), CGI(20), CGMU(20), EYAI(64), ETAJ(20), GJA(20), GM(20), SHJA(20), TFC(20), XNUI(64)
1 CCPJ(20), CEACT(64), CGI(20), CGMU(20), EYAI(64), ETAJ(20), GJA(20), GM(20), SHJA(20), TFC(20), XNUI(64)
2 CLN1MC(64), CLNPI(64), CMW(20), PGW(20), PI(64), PICHI(64), SENT(20), SHJ(20), TFB(20), THEV(20), XNUI(64)
3 GJB(20), PERTGJ(20), SAJ(20), SS(20), TFK(20), THEV(20), XNUI(64)
4 OO(64), SAJ(20), SS(20), TFK(20), THEV(20), XNUI(64)
5 SKIL(20), TFK(20), THEV(20), XNUI(64)
6 TFD(20), TFK(20), THEV(20), XNUI(64)
COMMON XNUIJ(64,20),XNUIJP(64,20)
1 COMMON IC, INEQ, INFOV, IP, IRUN, ISC, ISCP1,
2 ISMC, ISACNR, ISR, ISW, ISS, ISSP1, ISSP2, ISSP3, ISSP4,
3 ISVIA, ISW1A, ISW2A, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
4 ISVIA, ISW6A, ISW6B, IUPD, IZERO, IZJJK, LC, MI, NFIT,
5 NN, NNS, NOS, NOT, NTEST,
6 COMMON IGM(20), ITB(5), KUR(64,20), LPIJ(20,10)
COMMON ACOM, ELEMENT, HP
DIMENSION AAA(22,24), RTA(64,20), CAPO(31), CCI(20), D(20), GJ(20), SBL
: J(20), SDGJ(20), SHJAP(20), THEVP(20)
EQUIVALENCE (AAA(1,1),AAA(1,1)),(BETA(1,1),RTA(1,1)),
1 (BLBK(1),CAPO(1)),(CAPXTH(1),CCI(1)),(GJA(1),GJ(1)),
2 (CAPX(1),GJ(1)),(SAJ(1),SBJ(1)),(SS(1),SDCJ(1)),
3 (SHJA(1),SHJAP(1)),(THEV(1),THEVP(1))
COMMON /MODPAR/ XMPI,DXMP,FSTAG, CATFAC,TMODEL,XMOOP1,DXMOOP,
1 TPLATE,KDIM
COMMON /AREA/ ATPI(11,2),PARAM(3,12,2),RTHCM(2),NSECT(2),
1 COMMON /SECTU(2),ISHAPE(12,2),NPROFL(2),NPRFLS,NBL
COMMON /ROMAIN/ HS,SUPGO,MFITER
COMMON /MODPT/ TSDIAM(20),TSAR(20),NTS,MPL
COMMON /NEWMP/ FACMP,NMODBT
COMMON /RDLIST/ IGAS,IGASU,NOZZO,ICHAND
DIMENSION WORD(3,2)
DATA WORD/6HTWO-DI,6HMENSIO,3HNAL,6HAXIAL,6HY-SYMM,SHETRIC/
COMMON /LN/ ISATOM,ISMDL,JATOM,JMOL
COMMON /MASS/ SMAS,CTMXXI,TSTOPI,IS(20)
COMMON /BL/ DELHL(2),BLINT(2),XZERO,TWALL,CPWALL,VISROT,DIAM(2),
1 SW,RO,JDIM,IPCINT
COMMON /RENE/ GAMIN,HTEST,TETEST,OTEST,DCHLL,DCHRAT,CCHI,TTEST,
1 GTEST
COMMON /POLYAT/ THEVE(4,20)
COMMON /RDWEDG/ ANGLE(10),RADLE(5),WXI,DMX,WXI(20),TWEDGE,WK,
1 NWX,NANGLE,NRADLE,WEDGEM,AXISYM,ISW9B

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D J 140 I=1,ISKR
WRITE (6,570) I,KTR(I),KTR(I),ITR(I),EPAR(1,I),EPAR(2,I),IPA(I)
IF (NFIT) 160,190,160
WRITE (6,510)
D J 180 J=1,ISS
IF (IGJ(J)) 170,180,170
WRITE (6,520) HP(J),TFA(J),TFC(J),TFD(J),TFE(J),TFK(J),SHJAL
1P(J)
CONTINUE
IF (INEQV,NE,0) WRITE (6,580)
WRITE (6,590) HP(JATCM),HP(JMOL),(COM1(I,IAMBIP),I=1,2),(COM2(K,LE
1WIS),K=1,2)
IF (ISW3H,NE,0) GO TO 200
WRITE (6,530)
GO TO 210
WRITE (6,540)
IF (XMOOPI,GF,9.E19,AND,NTS,EQ,0) GO TO 340
WRITE (6,600)
CXMAXI=CXMAX/2,54
IF (XMOOPI,LT,9.E19) WRITE (6,630) NMOOPI,XMOOPI,CXMAXI
IF (NTS,EQ,0) GO TO 240
GO TO (220,230),NPFPLS
WRITE (6,610) (TSDIAM(I),I=1,NTS)
GO TO 240
WRITE (6,62) (TSDIAM(I),I=1,NTS)
IF (T,AXISYM) GO TO 290
WRITE (6,64) TMODEL,TPLATE
IF (FSIAG) 250,260,270
WRITE (6,650)
GO TO 280
WRITE (6,660)
GO TO 280
WRITE (6,670)
WRITE (6,680) CATFAC
KDPI=KDIMP+1
WRITE (6,690) (WORD(I,KDPI),I=1,3)
IF (.NOT.WEDGEM) GO TO 340
WRITE (6,700) (ANGLE(I),I=1,NANGLE)
WRITE (6,710) (RADLE(I),I=1,NRADLE)
IF (NW,EO,0) GO TO 300
WRITE (6,720) NWX,DWX,WX1
IF (WXI(I),GT,1.E19) GO TO 330
D J 310 I=1,20
IF (WXI(I),GT,1.E19) GO TO 320
NW,IXI=I
CONTINUE
WRITE (6,730) (WXI(I),I=1,NWXI)
WRITE (6,740) TWEDGE,WK
RETURN
FORMAT (1H,99H
1 ATOM FRACTION
1 FORMAT (1H,31X,A6,15X,1PE14.6,14X,E14.6)
ELEMENT
ELEMENT MOLECULAR WEIGHTS/

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IRAG COEFFICIENT = .FS.3)
END

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LIS 222-

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SUBROUTINE MATINV (A,MM,NN)
 MATRIX INVERSION ORIGINAL MATRIX
 INVERSE REPLACES A - ORIGINAL MATRIX
 CALLING SEQUENCE MM - SIZE TO BE INVERTED
 NN - FIRST SUBSCRIPT OF A IN DIMENSION
 STATEMENT OF MAIN PROGRAM

DOUBLE PRECISION A
 DATA RNAME /6HMATINV/
 DIMENSION A(NN,10),IV(10)
 M=MM

DO 10 I=1,M

IV(I)=I

JA=I

JJ=JA

K=JJ

KI=K+1

BIG=ABS(A(K,K))

IF (KI-M) 30,30,60

DO 50 I=KI,M

IF (BIG-ABS(A(I,JJ))) 40,50,50

K=I

BIG=ABS(A(I,JJ))

CONTINUE

BIG=A(K,JJ)

IF (BIG) 80,70,80

CONTINUE

WRITE (6,300)

CALL DUMP (RNAME)

RETURN

IF (K-JJ) 90,110,90

J=IV(K)

IV(K)=IV(JJ)

IV(JJ)=J

DO 100 J=1,M

B=A(JJ,J)

A(JJ,J)=A(K,J)

A(K,J)=B

A(JJ,JJ)=1.0

JA=JJ+1

DO 120 J=1,M

A(JJ,J)=A(JJ,J)/BIG

JB=JJ-1

IF (JB) 170,170,130

DO 160 I=1,JB

BB=A(I,JJ)

A(I,JJ)=0.0

IF (BB) 140,160,140

DO 150 J=1,M

A(I,J)=A(I,J)-BB*A(JJ,J)

CONTINUE

IF (JA-M) 180,180,220

DO 210 I=JA,M

BB=A(I,JJ)

A(I,JJ)=0.0

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190 IF (BB) 190,210,190
200 DO 200 J=1,M
210 A(I,J)=A(I,J)-BB*A(JJ,J)
220 CONTINUE
230 GO TO 20
240 M=M-1
250 IF (M) 290,290,230
260 DO 280 J=1,M1
270 DO 250 I=J,M
280 IF (IV(I)-J) 250,240,250
290 K=I
300 GO TO 260
310 CONTINUE
    WRITE (6,310)
    CALL DUMP (RNAME)
    RETURN
260 IV(K)=IV(J)
    DO 270 L=1,M
    AA=A(L,J)
    A(L,J)=A(L,K)
    A(L,K)=AA
    CONTINUE
    RETURN
    C
300 FORMAT (1H1,26H MATINV, MATRIX SINGULAR /)
310 FORMAT (1H1,37HINDEXING OR STORAGE FAILURE IN MATINV/)
    END

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MAT 83-

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EPS=SHV(2,ISOLN)
T2=SHV(1,ISOLN)
ICOUNT=0
LO=.FALSE.
HI=.FALSE.
IF (ISOLN.EQ.1) GO TO 70
DO 60 I=1,ISS
SAVEC(I)=GJ(I)*CM1
GO TO 110
P2=P1/J12*(1.-EPS)
ICOUNT=ICOUNT+1
IF (ICOUNT.GT.30) GO TO 450
CALL EQCALC (T2,P2)
IF (FRR) RETURN
EPSOLD=EPS
EPS=R1/ZRHO
IF (ABS(EPS/EPSCLD-1.).LE.0.001) GO TO 130
ALPHA=P2/ZRHO
DISC=PT**2-PC*ALPHA
IF (DISC.GE.0.) GO TO 100
EPS=EPSOLD
IF (LU) GO TO 90
T2=0.9*T2
GO TO 50
T2=0.5*(T2+TLO)
GO TO 50
P2=0.5*(PT+SQRT(DISC))
GO TO 80
ICOUNT=ICOUNT+1
CT=T2/CTAP
IF (ICOUNT.GT.30) GO TO 450
CALL THERM
ZCM=CM1
ZCH=0.
DO 120 I=1,ISS
ZCH=ZCH+SAVEC(I)*SHJ(I)
OP=P1*T2/T1
DISC=PT**2-4.*R1U12*OP
IF (DISC.LT.0.) GO TO 450
P2=0.5*(PT+SQRT(DISC))
R2=R1*P2/OP
EPS=R1/R2
H2=HS-HC*EPS**2
ZCH=ZCH+CMA/ZCM
IF (ABS(ZCH/H2-1.).LE.0.001) GO TO 170
ERZ=ZCH-H2
IF (ERZ) I4C,I4C,I50
FRRLO=ERZ
TLO=T2
LO=.TRUE.
IF (HI) GO TO 160
T2=1.1*T2
GO TO 50
ERRHI=ERZ
THI=T2

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MOD 111
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MOD 113
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MOD 164
MOD 165

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160 HI=.TRUE.
161 IF (LO) GO TO 160
162 T2=0.9*T2
163 GO TO 50
164 T2=T2-ERZ*(THI-TLO)/(ERRHI-ERRLO)
165 GO TO 50
166 SUBSONIC FLOW FROM THE SHOCK TO THE STAGNATION POINT ON THE MODEL
167 PS=F2+C.5*RIUI2*EPS
168 TS=T2
169 HI=.FALSE.
170 ICOUNT=0
171 ICOUNT=ICOUNT+1
172 IF (ICOUNT.GT.20) GO TO 450
173 GO TO (190,200).ISOLN
174 CALL EGCALC (TS,PS)
175 IF (ERR) RETURN
176 GO TO 220
177 CT=TS/CTAP
178 CALL THERM
179 ZCH=0.
180 DO 210 I=1,ISS
181 ZCH=ZCH+SAVEC(I)*SHJ(I)
182 ZCH=ZCH*CMA/ZCM
183 IF (ABS(ZCH/HS-1.)*LE.0.001) GO TO 260
184 ERZ=ZCH-HS
185 IF (ERZ) 230,230,240
186 TLO=TS
187 ERKLO=ERZ
188 IF (HI) GO TO 250
189 TS=1.1*TS
190 GO TO 180
191 THI=TS
192 ERRHI=ERZ
193 HI=.TRUE.
194 TS=TS-ERZ*(THI-TLO)/(ERRHI-ERRLO)
195 GO TO 180
196 IF (ISCLN.EQ.1) GO TO 280
197 IF (.NOT.AXISYM.OR.FSTAG.EQ.0.) GO TO 300
198 ZSENE=ALOG(PS)
199 DO 270 I=1,ISS
200 IF (SAVEC(I).LE.0.) GO TO 270
201 ZSENE=ZSENE+SAVEC(I)*(SENT(I)-ALOG(SAVEC(I)))
202 CONTINUE
203 ZRHO=2*PS/P2*T2/TS
204 VGP(1) = VELOCITY GRADIENT PARAMETER FOR HEMISPHERE MODEL
205 VGP(2) = VELOCITY GRADIENT PARAMETER FOR FLAT-FACED MODEL
206 SOD(1) = STANCOFF DISTANCE FOR HEMISPHERE (CYLINDER IF KDIM=0)
207 SOD(2) = STANCOFF DISTANCE FOR FLAT-FACED MODEL
208 VGO(1)=SQRT(2.*PS*CATM*(1.-PI/PS)/ZRH0)/UI
209 SOTE=SQRT(EPS)
210 IF (KDIM.EQ.0.) GO TO 290
211 SOD(1)=0.78*EPS
212 IF (EPS.GT.C.1901) SOD(1)=SOD(1)*(1.+3.5*(EPS-0.19)**1.6)
213 VCP(2)=EPS/(1.24+4.2*(SQTE-0.4)**2)
214 SOD(2)=1.12*EPS**C.39
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320 GO TO 300
321 SDD(1)=1./2./EPS/((1.+0.5*EPS)*ALOG(1.3333333/EPS))-0.5*EPS)-1.)
322 SDD(2)=-1.23*SQTE*ALOG(0.79*SQTE)
323 VGP(2)=0.64*EPS
324 SHV(1,ISOLN)=ET2
325 SHV(2,ISOLN)=EFS
326 IF (.NOT.AXISYM.OR.(ISCLN.EQ.2.AND.FSTAG.EQ.0.)) GO TO 440
327 IF (ISCLN.EQ.2) GO TO 320
328 DD 310 I=1,ISS
329 SAVEC(I)=ZCAP(I)
330 HCF=0.
331 IF (NOTRAN) GO TO 360
332 DD 340 I=1,ISS
333 NEL=0
334 DD 330 J=1,ISC
335 IF (LPIJ(I,J).EQ.0) GO TO 330
336 IF (LPIJ(I,J).NE.1) GO TO 340
337 NEL=NEL+1
338 IF (NEL.GT.1) GO TO 340
339 CONTINUE
340 HCF=HCF+SAVEC(I)*SHJA(I)
341 CONTINUE
342 HCF=HCF*CRP/ZCM
343 CALL TRANSP (TS,PS)
344 IF (ERR) RETURN
345 HSTAG=CRP/CMA*ZCH
346 H*=CP*ALL*TMDEL
347 HRATIO=HCF/(HSTAG-HW)
348 IF (LEWIS.EQ.2) HRATIO=0.
349 QSR1=0.0750*SQRT(PS)*(HSTAG-HW)
350 QSR2=1.8*0.0417. WHERE FACTOR 1.8 CONVERTS ENTHALPIES FROM CAL/GMM
351 TO BTU/LB
352 DD 350 L=1,2
353 QS=0.51/(PR**PR)**0.3*SQ2F(KDIM+1)*SQRT(ZRHO*VISC*UI*VGF(L))*(HSTAG
354 IG-HW)
355 QFRF(L)=QS*(1.+(FLEWIS**ELN(1))-1.)*HRATIO)
356 QFRF(L)=QS*(1.+(FLEWIS**ELN(2))*CATFAC-1.)*HRATIO)
357 CKSI=(EPS/(1.+EPS))*2*RI*UI/VISC*30.48
358 SCOUT(1)=TS
359 SCOUT(2)=CF(1)*ZRHO/RHAP
360 SCOUT(3)=PS
361 SCOUT(4)=VISC*CF(3)
362 SCOUT(5)=FLEWIS
363 SCOUT(6)=Z*FN*CRA/ZCM
364 SCOUT(7)=ZCM
365 SCOUT(8)=SIGMA
366 SCOUT(9)=PR
367 SCOUT(10)=EPS
368 IF (NOTRAN) GO TO 370
369 SCOUT(11)=QFRF(2)
370 SCOUT(12)=QFRF(2)
371 SCOUT(13)=QSR1
372 SCOUT(14)=SQD(2)
373 SCOUT(15)=SQD(1)
374 SCOUT(15)=QFRF(1)

```


50 ITHAN 1.5.)
 510 FFORMAT (80H0-----FROZEN PRANDTL NUMBER AT STAGNATION-POINT WAMOD 330
 520 1LL TEMPERATURE ON MODEL =,F7.4) MOD 331
 FFORMAT (46X,19HMODEL CONDITIONS -.3A6) MOD 333
 1.A2,4X,E10.3,4X,A6,A2,4X,1PE10.3,4X,A6,A2,4X,E10.3,4X,A6MOD 334
 25.A2,4X,1PE10.3,4X,A6,A2,CPF11.3/1X,A6,A2,OPF10.2,8X,A6,A2,F10.2,8X,AMCD 335
 3X).2(1X,A6,A2,F11.3,6X)/3(1X,A6,A2,F10.2,7X),1X,A6,A2,F11.3,7X,A6, MOD 336
 4A2,F10.2) MOD 337
 FFORMAT (46X,17HMOLE FRACTIONS ..,3A6) MOD 338
 540 FFORMAT (5(1X,A4,3X,1H=.4X,1PE10.3,3X)) MOD 340
 550 FFORMAT (57H0ITERATION TO FIND STAGNATION CONDITIONS DID NOT CONVERMOD 341
 1GE) MOD 342
 560 FFORMAT (46H0ELAPSED TIME IN MODEL AND WEDGE CALCULATIONS=,F4.0,8H MOD 343
 1SECONDS) MOD 344
 END MOD 345-

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60      NIT=0
70      M2=ISC+2
          ZP=ALOG(PRES)
          DO 90 J=IM,ISC
            AA(J,M2)=OM(J)-CAPX(J)
            IF (NODEPS) GO TO 90
          DO 80 I=1,ISMCNR
            AA(J,M2)=AA(J,M2)-(CDIJ(I,J)-OM(J)*BET(I))*SKIL(I)
          CONTINUE
          AA(M1,M2)=-ZP-ZPA
          DO 100 L=IM,ISSNR
            AA(M1,M2)=AA(M1,M2)+CAPX(L)*(PGJ(L)-CGMU(L))
          DO 150 J=IM,ISC
            DO 150 K=IM,ISC
              IF (J-K) 120,110,120
              GO TO 130
            AA(J,K)=0.0
            IF (NODEPS) GO TO 150
          DO 140 I=1,ISMCNR
            AA(J,K)=AA(J,K)+CDIJ(I,K)*SKIL(I)*(CDIJ(I,J)-OM(J)*BET(I))
          CONTINUE
          DO 170 K=IM,ISC
            AA(M1,K)=(1.-PGJ(K)+CGMU(K))*CAPX(K)
            IF (NODEPS) GO TO 170
          DO 160 I=1,ISMCNR
            L=I+ISC
            AA(M1,K)=AA(M1,K)+CDIJ(I,K)*(1.-PGJ(L)+CGMU(L))*CAPX(L)
          CONTINUE
          AA(M1,M1)=1.0
            IF (NODEPS) GO TO 190
          DO 180 I=1,ISMCNR
            L=I+ISC
            AA(M1,M1)=AA(M1,M1)+BET(I)*(1.-PGJ(L)+CGMU(L))*CAPX(L)
          DO 210 J=IM,ISC
            AA(J,M1)=0.0
            IF (NODEPS) GO TO 210
          DO 200 I=1,ISMCNR
            L=I+ISC
            AA(J,M1)=AA(J,M1)+BET(I)*CAPX(L)*(CDIJ(I,J)-OM(J)*BET(I))
          CONTINUE
          IF (JJK) 240,220,240
          IF (CAPX(I)-1.0D-20) 230,230,270
          JJK=1
          ISMCNR=ISMCNR-IC
          ISSNR=ISSNR-IC
          NODEPS=ISMCNR.EQ.0
          IM=IM+1
          AA(1,1)=1.0
          DO 250 J=2,M2
            AA(1,J)=0.0
            CAPX(I)=0.0
            LL=ISSNR+1
            DO 260 I=LL,ISS
              CAPX(I)=0.0

```

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NEW 56
NEW 57
NEW 58
NEW 59
NEW 60
NEW 61
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NEW 109
NEW 110

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2. CALL DSMSOL (AA,M1,22)
IF (ERR) RETURN
NIT=NIT+1
DO 340 K=IM,ISC
  ZB=1.+AA(K,M2)
  CAPX(K)=CAPX(K)/2.
  IF (CAPX(K)) 290,300,290
  CGMU(K)=ALOG(CAPX(K))
  GO TO 340
300 WRITE (6,480) K
  CGMU(K)=CGMU(K)-0.69314718
  GO TO 340
310 CAPX(K)=CAPX(K)*ZB
  IF (CAPX(K)) 320,330,320
320 CGMU(K)=ALOG(CAPX(K))
  GO TO 340
330 WRITE (6,480) K
  CGMU(K)=CGMU(K)+ALOG(ZB)
  CONTINUE
  ZC=1.+AA(M1,M2)
  IF (ZC) 350,350,360
  PRES=PRES/ZC.
350 IF (PRES) 370,380,370
  PRES=PRES*ZC
360 IF (PRES) 370,390,370
  ZP=ALOG(PRES)
  GO TO 400
380 WRITE (6,490)
  ZP=ZP-0.69314718
  GO TO 400
390 WRITE (6,490)
  ZP=ZP+ALOG(ZC)
  IF (NODEPS) GO TO 430
400 DO 420 I=1,ISMCNR
  SKIL(I)=CHI(I)+BET(I)*(ZP+ZPA)
  DO 410 L=IM,ISC
  SKIL(I)=SKIL(I)+CDIJ(I,L)*CGMU(L)
  J=I+ISC
  CGMU(J)=SKIL(I)
  SKIL(I)=EXP(SKIL(I))
  CAPX(J)=SKIL(I)
  CONTINUE
420 DO 450 K=IM,M1
430 IF (ABS(AA(K,M2)).LE.TEST) GO TO 450
  IF (NIT-NTEST) 70,70,440
440 WRITE (6,500)
  CALL DUMP (RNAME)
  RETURN
450 CM=0.0
  CM=0.0
  DO 460 I=IM,ISSNR
  CM=CM+CAPX(I)*CGI(I)
  CH=CH+CAPX(I)*SFJ(I)
460

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NEW 164
NEW 165

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RHOBAR=PRES*CM/(CMA*CT)
RHO=(RHOBAR*RUBARA)/(1.+RHOBAR*ROBARP*BZERO)
CH=CH*(CMA/CM)
CH=CH+(PRES/RHOBAR)*(RHOBAR*(ROBARA/RHO)-1.)
DO 470 I=1M,ISSNR
  GJ(I)=CAPX(I)/CM
SU2=2.*(CHA-CH)
SU=SQRT(SU2)
FLUX=RHO*SU
RETURN
      FORMAT (16H IN NEWRAP,CAPX(.12,3H)=0)
      FORMAT (14H IN NEWRAP,P=0)
      FORMAT (35H100 MANY NEWTON-RAPHSON ITERATIONS)
      END

```

470

C
C
C

480
490
500

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NEW 166
NEW 167
NEW 168
NEW 169
NEW 170
NEW 171
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NEW 173
NEW 174
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NEW 176
NEW 177
NEW 178
NEW 179
NEW 180
NEW 181
NEW 182-

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1 NEX
 2 NEX
 3 NEX
 4 NEX
 5 NEX
 6 NEX
 7 NEX
 8 NEX
 9 NEX
 10 NEX
 11 NEX
 12 NEX
 13 NEX
 14 NEX
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 16 NEX
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 18 NEX
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 21 NEX
 22 NEX
 23 NEX
 24 NEX
 25 NEX
 26 NEX
 27 NEX
 28- NEX

```

SUBROUTINE NEXTMP (ITS,XMOD1,XMODEL)
LOGICAL ERR
COMMON /ERROR/ ERR
COMMON /AREA/ ATPI(11,2),PARAM(3,12,2),RTHCM(2),NSECT(2),
1 NSECTU(2),ISHAPE(12,2),NPROFL(2),NPRFLS,NBL
COMMON /NEWMP/ FACMP,NMODPT
COMMON /MODPT/ TSDIAM(20),TSAR(20),NTS,MBL
COMMON /MODPAR/ XMPI,DXMP,FSTAG, CATFAC,TMODEL,XMODP1,DXMODP,
1 TPLATE,KDIM
DATA ONE /1./
IF (NPRFLS.EQ.1) GO TO 20
IF (.TS.LE.NTS) GO TO 10
XMODEL=1.E30
GO TO 50
10 CALL FINDXC (TSAR(ITS),MBL,XMOD2)
IF (ERR) RETURN
GO TO 40
20 IF (ITS.GT.NTS) GO TO 30
CALL FINDX (TSAR(ITS),ONE,XMOD2)
IF (ERR) RETURN
IF (XMOD1-XMOD2) 30,30,40
XMODEL=XMOD1
XMOD1=XMOD1*FACMP
GO TO 50
40 XMODEL=XMOD2
ITS=ITS+1
RETURN
END
  
```

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SUBROUTINE NONEQ
REAL ACOM(30),ELMENT(10),HP(20)
LOGICAL ERR,MODLPT,XMSET,FAILED,COUPLD,AXISYM,WEDGEM,NOREAC,DATAPEN
DOUBLE PRECISION AA,AAA,CJ,GJ,CAPX,DCI,CI,GSQ,DCA,AIN,CDIJ,ONE
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A,AFNTS,AFNX,AMACH,AR,ARBA,ARBB,ARCB,BZERO,
C,CARB,CRS,CSTA,CT,CTAP,CM,CMA,CRA,CRP,
1 CRRB,CRS,CSTA,CT,CTAP,CM,CMA,CRA,CRP,
2 CTP,CTPL,CTT,CX,CXB,CXMAX,DATEST,OBTEST,DELT1,
3 DELT2,DELTA,CLOGA,DLOGR,DT,ENT,FLUX,HDELX,PCT,
4 SPTTEST,PRES,PFESA,PRESB,PRESTH,PRHO,RHAP,RHO,PHOB,
5 ERHOBAR,RHOC,RHOP,RHPL,RHTR,ROBARA,ROBARP,SCPG,SDT,
6 SEN,SHPG,SC,SL,SL64,SM,SU,SU2,SUMG,
7 TEST,TESTB,TPRINT,TSTOP,UP,ZP,ZPA
8 COMMON BE(64),BET(20),DLBK(31),CAI(64),CAPXTH(20),NON
9 CCPJ(20),CEACT(64),CGI(20),CGMU(20),CHI(64),CHI(20),NON
10 CLNIMC(64),CLNPI(64),CMW(20),ETAI(64),ETAJ(20),NON
11 GJB(20),PERTGJ(20),PGJ(20),PI(64),PICH(64),GM(20),NON
12 GO(64),SAJ(20),SDCHI(64),SENT(20),SHJI(20),NON
13 SKIL(20),SS(20),TB(30),TFA(20),TFB(20),NON
14 TFD(20),TFE(20),TFK(20),THEV(20),XMJAT(20),NON
15 COMMON BETA(64,20),ELJ(10,20),GELJ(10,20),NON
16 XNUIJ(64,20),XNUIJP(64,20)
COMMON IC,IM,INEQ,INEQV,IP,IRUN,ISC,ISCP1,
1 ISMC,ISMCMR,ISR,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
2 ISW1A,ISW1B,ISW2A,ISW2B,ISW3A,ISW3B,ISW4A,ISW4B,ISW5A,
3 ISW5B,ISW6A,ISW6B,IUPD,IZERO,JJK,LC,M1,NFIT,
4 NIT,NNN,NNS,NQS,NOT,NTEST
COMMON IGM(20),ITB(5),KUR(64,20),LPIJ(20,10)
COMMON ACOM,ELMENT,HP
DIMENSION AAA(22,24),BTA(64,20),CAPQ(31),CCI(20),DGJ(20),SBMON
1J(20),SDGJ(20),SHJAP(20),THEVP(20)
EQUIVALENCE (AA(1,1),AAA(1,1)),(BTA(1,1),BTA(1,1)),
1 (BLRK(1),CAPQ(1)),(CAPXTH(1),CCI(1)),(GJA(1),DGJ(1)),
2 (CAPX(1),GJ(1)),(SAJ(1),SBJ(1)),(SS(1),SDGJ(1)),
3 (SHJA(1),SHJAP(1)),(THEV(1),THEVP(1))
DIMENSION CI(10),DCI(10),DCA(10),GSQ(20)
EQUIVALENCE (GSQ(1),AA(1,24))
EQUIVALENCE (CCI(1),AA(1,23)),(DCA(1),AA(11,23))
COMMON /TAPOUT/ XXX(10),ITPOUT,NRCOUT,IFLOW,ITYPER,IMP,DATAPEN
COMMON /INGNE/ AIN(10,10)
COMMON /BL/ DELBL(2),BLINT(2),XZERO,TWALL,CPWALL,VISROT,DIAM(2),
1 SW,RO,JDIM,IPCINT
COMMON /NEQ/ OMDST(2),DDELBL(2)
COMMON /MODPT/ TSDIAM(20),TSAR(20),NTS,MBL
COMMON /ERROR/ ERR
COMMON /CERROR/ FAILED
COMMON /BLNE/ XI(2),ORDINP(2),DELBLP(2),DBLP(2),AMPP,AMP,
1 COUPLD,ISMDO
COMMON /HLOUT/ REPF,THETA(2),SN(2),XSN(2),PREF,HR,QWDDOT(2),
1 TAUW(2)
DIMENSION BLNEAR(10),ISUPR(64),CAISV(64)
EQUIVALENCE (BLNEAR(1),XI(1)),(ISUPR(1),AA(1,1))
COMMON /RENE/ GAMIN,HTEST,TETEST,OTEST,DHLL,DCHRAT,CCHI,TTEST,
1 GTEST

1 NON
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3 NON
4 NON
5 NON
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46 NON
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52 NON
53 NON
54 NON
55 NON


```

SCD=2*
ITB(I)=0
NNS=0
NIN=0
ISSP1=ISS+1
ISSP2=ISS+2
ISSP3=ISS+3
ISSP4=ISS+4
INEQ=0
DO 30 I=1,ISR
PI(I)=1.0
ZZZ=CAI(I)
CAI(I)=ALOG(ZZZ)-ETAI(I)*ALOGT
CAISV(I)=CAI(I)
BE(I)=0.0
XNUI(I)=0.0
DO 30 J=1,ISS
BETA(I,J)=XNUIJP(I,J)-XNUIJ(I,J)
BE(I)=BE(I)+BETA(I,J)
XNUI(I)=XNUI(I)+XNUIJ(I,J)
ZZZ=CMA/CRP
CSTA=.5*ALOG(ZZZ)
SL64=SL/6468.
COMPUTE AND TEST RANK OF BETA MATRIX
IX=0
I=0
I2=1
I=I+1
IF (I-ISS) 50,50,150
DO 70 J=I2,ISR
IF (BETA(J,I)) 60,70,60
KA=J
IX=IX+1
GO TO 80
CONTINUE
GO TO 40
K=KA
DO 90 M=1,ISS
AB=BETA(K,M)
BETA(K,M)=BETA(IX,M)
BETA(IX,M)=AB
I2=IX+1
IF (I2-ISS) 100,100,150
DO 120 N=I2,ISR
CA=BETA(I,N)/BETA(IX,I)
IF (CA) 110,140,110
DO 130 N=I,ISS
BETA(M,N)=BETA(M,N)-CA*BETA(IX,N)
IF (ABS(BETA(M,N))-0.01) 120,120,130
BETA(I,N)=0.0
CONTINUE
CONTINUE
GO TO 40
IF (IX-ISS) 160,170,160

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NON 162
NON 163
NON 164
NON 165

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160 WRITE (6,1090)
C   ERROR EXIT NO. 1 *****
C   BETA MATRIX OF INCORRECT RANK *****
C   IERR=1 *****
C   GO TO 1000
C   RESET BETA MATRIX (DESTROYED DURING RANK CALCULATION)
170 DO I=1,ISR
    DO J=1,ISS
180   BETA(I,J)=XNUIJP(I,J)-XNUIJ(I,J)
      AR=C.0
      LC=0
      DELT=DELT1
      DO 200 I=1,ISC
        CI(I)=0
190   DO J=1,ISS
        C(I)=CI(I)+GJA(J)*LPIJ(J,I)
200   CONTINUE
      GO TO 220
C   END OF INITIALIZATIONS
210 IF (MODLPT) CALL NEXTMP (ITS,XMOD1,XMODEL)
220 CALL DERIVS
      IF (ERR) RETURN
      IF (.NOT.FAILED) GO TO 230
C   ERROR EXIT NO. 2 *****
C   IERR=2 *****
230 GO TO 1000
      IF (XMSF.OR.AMACH.LT.1.5) GO TO 240
      XMSET=.TRUE.
      XMOD1=XMPI
      ITS=1
240 CALL NEXTMP (ITS,XMOD1,XMODEL)
      IF (IUPD) 250,250,250
C   TESTS FOR SWITCH TO DOWNSTREAM REGION
250 IF (OLOGA) 450,260,260
260 IF (CX) 450,450,270
270 IF (AR) 300,280,300
280 AR=1.0
      DO 290 I=1,NPRELS
        OMDST(I)=1.-DELBL(I)
290 IF (AFNX.LT.DATEST) GO TO 450
300 SAVE DATA AT SWITCH POINT FOR POSSIBLE RESTART
C   TB(1)=CX
      TB(2)=CT
      TB(3)=SC
      TB(4)=CTT
      TB(5)=DELTAX
      TB(6)=HDELX
      TB(7)=CTE
      TB(8)=SUD
      TB(9)=CHA
      ITB(1)=INEQ
      ITB(2)=NNS
      ITB(3)=NNN
      ITB(4)=NRRCOUT

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310 DO 310 I=1,ISS
    TB(I+10)=GJ(I)
    BLBK(I)=RSA
    IF ((ISW3B.EQ.0) GO TO 340
DO 320 I=1,10
    BLBK(I+1)=BLNEAR(I)
    BLBK(I+2)=XPB
    BLBK(I+3)=DXPB
DO 330 I=1,NPRFLS
    BLBK(I+13)=BLINT(I)
    BLBK(I+15)=XSN(I)
    BLBK(I+17)=DELBL(I)
    BLBK(I+19)=DDELBL(I)
330 SWITCH TO DOWNSTREAM REGION
C 340 CALL THROAT
    IF (ERR) RETURN
    GO TO 450
350 IF (DLOGR) 450,450,360
360 WRITE (6,1100)
370 IF ((AFNX-DATEST)/DATEST.LT.0.05) GO TO 380
C 380 ERROR EXIT NO. 3 *****
C      POSITIVE DLOGR FAR BEYOND SWITCH POINT
    IERR=3
    GO TO 1000
C      SWITCH TO DOWNSTREAM REGION WAS PREMATURE.
C      INCREASE DATEST AND RESTART UPSTREAM SOLUTION AT SWITCH POINT.
380 DATEST=2.*(DATEST-1.))+1.
    IUPD=1
    ARBB=ARBB+1.0
    IF (ARBB.LT.ARBA) GO TO 390
C      ERROR EXIT NO. 4 *****
C      IERR=4
    GO TO 1000
390 TSZ=TPRINT
    TPRINT=0.0
    CALL PRTA
    IF (ERR) RETURN
    ITAPE=NRCONT-ITB(4)
    IF (ITAPE.EQ.0) GO TO 410
C      BACKSPACE BINARY OUTPUT TAPE TO ELIMINATE RECORDS OF INVALID
C      STEPS BEYOND SWITCH POINT
    DO 400 I=1,ITAPE
    BACKSPACE ITPOUT
400 TPRINT=TSZ
410 RESET DATA FOR RESTART AT SWITCH POINT
    CX=TB(1)
    CT=TB(2)
    SC=TB(3)
    CTT=TB(4)
    DELTAX=TB(5)
    HDELX=TB(6)
    CTE=TB(7)
    SCD=TB(8)
    CHA=TB(9)
    INEQ=ITB(1)

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420 NNS=ITB(2)
      NNN=ITB(3)
      NRCOUT=ITB(4)
      DO 420 I=1,ISS
      GJ(I)=TB(I+10)
      RSA=BLBK(I)
      IF (ISW3B.EQ.0) GO TO 220
      DO 430 I=1,10
      BLNEAR(I)=BLBK(I+1)
      XPC=BLBK(I2)
      DXPB=BLBK(I3)
      DO 440 I=1,NPRFLS
      BLINT(I)=BLBK(I+13)
      XSN(I)=BLBK(I+15)
      DELBL(I)=BLBK(I+17)
      DDELBL(I)=BLBK(I+19)
      ISMD=30
      COUPLD=.FALSE.
      GO TO 220
430 CONTINUE
      IF (INEG) 610,460,610
      PERTURBATION SOLUTION
      CALL PERT
      IF (ERR) RETURN
      IF (IUPD.EQ.C.AND.DELT.EQ.DELT1) DELT=DELT5V
      DETERMINE MAXIMUM AND MINIMUM PERTURBATIONS IN CHI(I)
      DCHMAX=C.
      DCHMIN=1.E30
      DO 480 I=1,ISR
      ADCH=ABS(SDCHI(I))
      IF (ADCH-DCHMAX) 480,480,470
      DCHMAX=ADCH
      IMAX=I
      DCHMIN=AMINI(CC IN,ADCH)
      WRITE (6,1110) DCHMIN,DCHMAX,IMAX
      IF (DCHMAX.EQ.0.) GO TO 580
      RATIO=DCHMIN/DCHMAX
      IF (RATIO.GE.DCHPRAT) GO TO 510
      EXCESSIVELY SMALL RATIO OF MINIMUM AND MAXIMUM PERTURBATIONS
      IN CHI(I)
      DO 500 J=1,ISS
      IF (BETA(IMAX,J)*PICH1(IMAX).GE.0.) GO TO 500
      IF (GJ(J).GT.GAMIN) GO TO 500
      IF (IRA.LE.50) GO TO 490
      WRITE (6,1010)
      GO TO 530
      ARTIFICIAL INCREASE OF RATE CONSTANT TO PREVENT PREMATURE START
      OF INTEGRATION
      RATIO2=1.I*DCHPRAT/RATIO
      CAI(IMAX)=CAI(IMAX)+ALOG(RATIO2)
      WRITE (IRA+1
      GO TO 220
      CONTINUE
      TESTS FOR SWITCH FROM PERTURBATION SOLUTION TO INTEGRATION
500
C

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510 IF (DCHMAX.LT.CCHI) GO TO 580
C IF (DCHMAX.LE.PCTEST*CCHI) GO TO 540
C MAXIMUM SDCHI(I) IS TOO LARGE -- BACKSTEP THE PERTURBATION
  SOLUTION TO FIND THE CORRECT SWITCHING POINT
  DELT=DELT/2.0
  ISUBD=ISUBD+1
  IF (ISURD.GT.23) GO TO 520
  CT=CT+DELT
  CTE=CT
  NAMELIST /NEODMP/ INEQ,IURD,DLOGA,AR,CX,CT,DELT,AFNTS,
1  UP,SDCHI,ODELBL,PCT,PRHO,GJ,PERTGJ
  IF (ISWSB.NE.0) WRITE (6,NEQDMP)
  GO TO 600
520 WRITE (6,1120) ISUBC
530 WRITE (6,NEQDMP)
C ERROR EXIT NO. 5 *****
IERR=5
GO TO 1000
C SWITCH FROM PERTURBATION SOLUTION TO INTEGRATION
540 INEQ=1
  DXM=0.01*DCHMIN/DCHLL
  IF (DELTAX.LE.DXM) GO TO 550
  DELTAX=DXM
  HDELX=0.5*DELTAX
  TB(5)=DELTAX
  TB(6)=HDELX
  DXOLD=DELTAX
  CT=CT+PCT
  CTE=CT
  RHO=RH0+PRHO
  DO 560 I=1,ISR
  IF (CAI(I).EQ.CAISV(I)) GO TO 560
  CAI(I)=CAISV(I)
  WRITE (6,1030) I
  CONTINUE
560 DO 570 J=1,ISS
570 GJ(J)=GJ(J)+PERTGJ(J)
  CALL DERIV
  IF (ERR) RETURN
  IF (.NOT.FAILED) GO TO 610
  ERROR EXIT NO. 6 *****
  IERR=6
  GO TO 1000
C CONTINUE PERTURBATION SOLUTION
580 IF (CX.LT.XMODEL) GO TO 590
  MODLPT=.TRUE.
  CALL NEXTMP (ITS,XMOD1,XMODEL)
  PRINT POINT IN PERTURBATION SOLUTION
C CALL PRTA
  IF (ERR) RETURN
  IF (CX.GE.CXMAX) GO TO 660
  IF (CT.LE.TSTCP) GO TO 660
  ISUBD=J
  CT=CT-DELT
  CTE=CT
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66 CALL NEWRAP
   IF (ERR) RETURN
   AFNTS=SM/FLUX
   CALL AXFIT
   IF (ERR) RETURN
   GO TO 220
C   NUMERICAL INTEGRATION
510 IF (CT-1.0) 630,630,620
520 WRITE (6,1130)
C   ERROR EXIT NO. 7 *****
   IERR=7
   GO TO 1000
630 IF (CX.GE.CXMAX) GO TO 660
   IF (CT.LE.TSTOP) GO TO 660
   NN=NN+1
C   PRINT POINT IN SOLUTION BY INTEGRATION
   CALL PRTA
640 IF (ERR) RETURN
   IF (NN-NQS) 670,640,640
   DELTAX=SC*DELTAX
   HDELX=DELTAX/2.0
   NNS=NNS+1
   NN=0
650 IF (NNS-NQS) 670,650,650
   SC=SC+0.1
   SC=SC
   NNS=0
   GO TO 670
C   FINAL POINT
660 TPRINT=0.0
   CALL PRTA
   AXISYME=.TRUE.
   CALL MDEL
   WRITE (6,1160) ISTEPS
C   MAIN RETURN STATEMENT.....
C
C   RETURN
C   SAVE DATA AT BEGINNING OF INTEGRATION STEP FOR POSSIBLE RESTART
670 DO 680 I=1,ISS
   GJB(I)=GJ(I)
680 CONTINUE
   CXB=CX
   CTB=CT
   CARB=APX
   RHOB=RHO
   CRB=SEN
   CHRECHA
   DELBLB(1)=DELBL(1)
   IF (NPR.S.EQ.2) DELBLB(2)=DELBL(2)
   IF (CX+DELTAX.LT.XWDEL) GO TO 700
   MDELPT=.TRUE.
   DXSAVE=DELTAX
   HDXSAVE=HDELX
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700 DELTAX=XMODEL-CX
      HDELX=0.5*DELTAX
      IF (CX+DELTAX.LT.CXMAX) GO TO 710
      DELTAX=CXMAX-CX
      HDELX=0.5*DELTAX
      INTEGRATION STEP
      CALL RNKT RETURN
      IF (ERR) RETURN
      IF (FAILED) GO TO 740
      IF (.NOT.MODLPT) GO TO 720
      DELTAX=DXSAVE
      HDELX=HDXSAV
      CONTINUE
720 C TESTS FOR VALID INTEGRATION STEP
      IF FAIL=1
        IF (ABS(SDT/CT).GT.YTEST) GO TO 740
        IF (INT.EQ.1) GO TO 730
      IF FAIL=2
        IF (ABS(SDTE/CTE).GT.YTEST) GO TO 740
      IF FAIL=3
        IF (ABS(SDCHA/CHA).GT.HTEST) GO TO 740
      IF FAIL=4
        IF (LOGA*CX.GE.-0.01) GO TO 760
      IF (INTEGRATION STEP FAILED. REDUCE STEP SIZE AND RESTART AT
      BEGINNING OF STEP
730 C DELTAX=DELTAX/SCD
      HDELX=HDELX/2.0
      MODLPT=.FALSE.
      ICOUNT=ICOUNT+1
      IF (ICOUNT.GT.30.OR.DELTAX.LE.1.E-10) GO TO 990
      IF (YCOUNT) =IFAIL
      DO 750 I=1,ISS
      GJ(I)=GJR(I)
      CTE=CTEB
      CTE=CTEB
      CX=CXB
      AFNX=CARB
      RHO=RHOB
      SEN=CRS
      CHA=CHB
      DELBL(1)=DELBLB(1)
      IF (NFRFLS.EQ.2) DELBL(2)=DELBLB(2)
      NNN=0
      NNS=0
      SC=SC-0.1
      SC=AMAX1(SC,1.1)
      SCD=1.1*SCD
      CALL DERIVS
      IF (ERR) RETURN
      IF (.NOT.FAILED) GO TO 690
      ERROR EXIT NO. 8 *****
      IF I=8
740 C GO TO 1000
      FURTHER VALIDITY CHECKS ON STEP
      DO 790 I=1,ISS

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770      IFAIL=I+10
          GJ(I)=GJB(I)+SDGJ(I)
          IF (GJ(I).LT.0.) GO TO 770
          VF (ABS(SDGJ(I)/GJB(I)).LE.GTEST) GO TO 790
          IF (GJB(I).GE.GAMIN) GO TO 740
          IF (SDGJ(I).LE.0.) GO TO 740
          SUPPRESS REACTIONS INVOLVING STEP-SIZE CONTROLLING SPECIES
          WITH CONCENTRATION BELOW GAMIN AND FALLING
          NSUPR=0
          DO 780 K=1,ISR
            IF (BETA(K,I).EQ.0.) GO TO 780
            NDREAC(K)=TRUE
            NSUPR=NSUPR+1
            ISUPR(NSUPR)=K
            CONTINUE
          WRITE (6,1053) HP(I),GJB(I),(ISUPR(K),K=1,NSUPR)
          GO TO 740
          VALID STEP. APPLY INCREMENTS TO DEPENDENT VARIABLES.
          CONTINUE
          CI=CTB+SDT
          IF (NT.EQ.1) GO TO 800
          CTE=CTEB+SDTE
          CHA=CHB+SDCHA
          RESTORE CONSERVATION OF CHEMICAL ELEMENTS
          DO 820 J=1,ISC
            CJ=0.
            DO 810 I=1,ISS
              CJ=CJ+GJ(I)*LPIJ(I,J)
            DCI(J)=CI(J)-CJ
            DO 830 I=1,ISS
              GSO(I)=GJ(I)*2
            IF (GSO(I).GT.9.) OR (I.GT.ISC) GO TO 830
            WRITE (6,1060) I
            ERROR EXIT NO. 9 *****
            IERR=9
            GO TO 1000
          CONTINUE
          ISMCP1=ISMCP1+1
          DO 850 K=1,ISC
            DCA(K)=0.
            DO 840 J=1,ISC
              DCA(K)=DCA(K)+DCI(J)*AIN(J,K)
            CONTINUE
          DO 890 I=1,ISMCP1
            II=I+ISC
            AA(I,ISMCP1)=0.
            DO 860 K=1,ISC
              AA(I,ISMCP1)=AA(I,ISMCP1)+CDIJ(I,K)*DCA(K)
            AA(I,ISMCP1)=AA(I,ISMCP1)*GSO(II)
            DO 880 L=1,ISMCP1
              AA(I,L)=0.
            DO 870 K=1,ISC
              AA(I,L)=AA(I,L)+CDIJ(I,K)*CDIJ(L,K)/GSO(K)
              AA(I,L)=AA(I,L)*GSO(II)

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880 IF (I.EQ.0) AA(I,L)=AA(I,L)+ONE
890 CONTINUE
    CALL DSM SOL (AA, ISMC, 22)
    IF (ERR) RETURN
    DO 900 I=1, ISMC
      II=I+ISC
      GJ(II)=GJ(II)+AA(I, ISMCP1)
      DO 930 K=1, ISC
        GJ(K)=0.
        DO 910 J=1, ISC
          GJ(K)=GJ(K)+CI(J)*AIN(J,K)
        DO 920 L=1, ISMC
          I=L+ISC
          GJ(K)=GJ(K)-GJ(I)*CDIJ(L,K)
        CONTINUE
        DO 940 I=1, ISS
          IFAIL=I+30
          IF (GJ(I)) 740, 940, 940
        CONTINUE
        COMPUTE SUPPLEMENTARY FLOW VARIABLES AND DERIVATIVES AT END OF
        C STEP (AND BEGINNING OF NEXT STEP)
        CALL DERIVS
        IF (ERR) RETURN
        IFAIL=60
        IF (IFAIL) GO TO 740
        IF (NT.EQ.1) GO TO 960
        C FINAL VALIDITY CHECK IN ELECTRONIC NONEQUILIBRIUM MODEL ---
        C FINAL TEST ON CHANGE IN ENERGY TRANSFER TO ELECTRONS
        IF (OECLJ.EQ.0.) GO TO 950
        IFAIL=5
        DQEM=AMAX1(QTEST*ABS(OEOLD), DQMAX)
        IF (ABS(QDPE-OEOLD).GT.DQEM) GO TO 740
        DQMAX=AMAX1(DQMAX, DQEM)
        OEOLD=GDPE
        FULLY SUCCESSFUL STEP. SET UP FOR NEXT STEP.
        SCDE=SC
        TPECT*CTAP
        TPE=CTE*CTAP
        IF (ISS.GE.0) GO TO 970
        WRITE (6, 1140) CX, DELTAX, TP, TEP, CHA, GDPE, ICOUNT
        IF (ICOUNT.EQ.0) GO TO 980
        WRITE (6, 1040) (IF(K), K=1, ICOUNT)
        ICOUNT=0
        ISTEPS=ISTEPS+1
        DELTAX=AMAX1(DELTAX, 0.7*DXOLD)
        DXOLD=DELTAX
        GO TO 210
        WRITE (6, 1150)
        ERROR EXIT NO. 10*****
        IERR=10
        IF (IUPD.EQ.0) GO TO 370
        WRITE (6, 1070) IERR
        CALL DUMP (RNAME)
        RETURN
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1010	FORMAT (71HOMORE THAN 50 RATE-CONSTANT INCREASES REQUIRED IN PERTUNION SOLUTION)	NON 606
1020	FORMAT (20H0====)CHMIN/DCHMAX=.1PE10.3.4IH IS TOO SMALL. CONCENTRION OF REACTANT .A4.16H IN REACTION NO..I3.3H IS.E10.3.8HMOLE/GMNON 607	NON 608
1030	2./36H ====RATE CONSTANT FOR PERTURBATION NO..I3.25H INCREASED BY A FACTOR OF.E10.3.27H FOR PERTURBATION SOLUTION./) NON 609	NON 610
1040	FORMAT (36H ====RATE CONSTANT FOR REACTION NO..I3.27H RESTORED TO CORRECT VALUE.) NON 611	NON 612
1050	FORMAT (24H CAUSES OF STEP FAILURES.20I4) NON 613	NON 614
1060	FORMAT (28H0.....CONCENTRATION OF .A4.10H FROZEN. AT.1PE10.3.9NON 615	NON 616
1070	1H MOLE/GM./11X.2CHSUPPRESSED REACTIONS.34I3/30X.30I3) NON 617	NON 618
1080	FORMAT (4H0GJ(.I2.16H)*2 UNDERFLOWED) NON 619	NON 620
1090	FORMAT (15H0ERRCR EXIT NO..I3.11H FROM NONEQ) NON 621	NON 622
1100	FORMAT (24H1NONEQUILIBRIUM SOLUTION//) NON 623	NON 624
1110	FORMAT (33H1BETA MATRIX OF INSUFFICIENT RANK) NON 625	NON 626
1120	FORMAT (18H 0LGR IS POSITIVE) NON 627	NON 628
1130	FORMAT (8H DCHMIN=.1PE10.3.10X.7HDCMAX=.E10.3.10X.5HIMAX=.I2) NON 629	NON 630
1140	FORMAT (55H0BACKSTEPPING OF PERTURBATION SOLUTION TERMINATED AFTER 1. I3.31H STEPS. DIAGNOSTIC DATA FOLLOW.) NON 631	NON 632
1150	FORMAT (41H1TEMPERATURE GREATER THAN RESERVOIR VALUE) NON 633	NON 634
1160	FORMAT (3H0X=.1PE12.5.5X.7HDELTA=.E10.3.5X.2HT=.OPF8.2.5X.4HTEP=.NON 635-	NON 636
	1F8.2.5X.4HCHA=.1PE15.7.5X.5HQDPE=.E10.3.5X.7HICOUNT=.I2) NON 637	NON 638
	FORMAT (60H0NONEQUILIBRIUM INTEGRATION UNABLE TO TAKE A CONVERGENT NON 639	NON 640
	1 STEP) NON 641	NON 642
	FORMAT (33H0NONEQUILIBRIUM SOLUTION REQUIRED.15.18H INTEGRATION STNON 643	NON 644
	1EPS) NON 645	NON 646
	END NON 647	NON 648

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SUBROUTINE NRMAX
LOGICAL ERR
REAL ACOM(30),ELMENT(10),HP(20)
DOUBLE PRECISION AA,AAA,CAPX,GJ,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A,AFNLS,AFN,AMACH,
C,CARB,RS,CSTA,CH,
CTP,CLNT,CTAP,CTC,CTM,
4 DELT2,DELTA,C,L,DLOGA,DLOGR,DT,CXB,CXMAX,DATEST,DBTEST,DELT,
5 SPCTEST,PRES,PRESA,PRESTH,PRHO,PRHAP,RHO,RHOB,
6 GRHUPAR,RHOC,RHOP,RHPL,RHTR,ROBARA,ROBARP,SCPG,SDT,
7 SEN,SHPG,SC,SL,SL64,SU,SU2,SUMG,
8 TEST,TESTB,TPRINT,TSTOP,ZP,ZPA
COMMON BE(1),BLBK(31),CAI(64),CAPXTH(20),
1 CCPJ(20),CEACT(64),CGI(20),CGMU(20),CHI(64),CHII(20),
2 CLN1MC(64),CLNPI(64),CMW(20),EYAI(64),ETAJ(20),GJA(20),
3 GJB(20),PERTGJ(20),PGJ(20),PI(64),PICHJ(64),QM(20),
4 QQ(64),SAJ(20),SDCHI(64),SENT(20),SHJ(20),SHJA(20),
5 SKIL(20),SS(20),TB(30),TFA(20),TFB(20),TFCC(20),
6 TFD(20),TFE(20),YFK(20),THEV(20),XNJAT(20),XNUI(64)
COMMON IC,IM,INEQ,INEQV,IP,IRUN,ISC,ISCP1,
1 ISMC,ISMCNR,ISR,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
2 ISW1A,ISW1B,ISW2A,ISW2B,ISW3A,ISW3B,ISW4A,ISW4B,ISW5A,
3 ISW5B,ISW6A,ISW6B,IUPD,IZERO,JJK,LC,MI,NFIT,
4 NIT,NNN,NNS,NOS,NQT,NTEST
COMMON IGM(20),ITB(5),KUR(64,20),LPIJ(20,10)
COMMON ACOM,ELMENT,HP
DIMENSION AAA(22,24),SHJAP(20),THEVP(20),SB
1J(20),SDGJ(20),SHJAP(20),CCCI(31),CCI(20),DGJ(20),GJ(20),
EQUIVALENCE (AA(1,1),AAA(1,1)),(BETA(1,1),BTA(1,1)),
1 (BLBK(1),CAPQ(1)),(CAPXTH(1),CCI(1)),(GJA(1),DGJ(1)),
2 (CAPX(1),GJ(1)),(SAJ(1),SBJ(1)),(SS(1),SDGJ(1)),
3 (SHJA(1),SHJAP(1)),(THEV(1),THEVP(1))
COMMON /ERROR/ ERR
MI=ISC+1
DELT=DELT1
PRES=1.0
TI=1.0
CT=TI
DO 10 J=1,ISS
GJ(J)=GJA(J)
F1=0.
CT=CT-DELT
T2=CT
CALL NEWRAP
IF (ERR) RETURN
F2=FLUX
CT=CT-DELT
T3=CT
CALL NEWRAP
IF (ERR) RETURN
F3=FLUX

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SUBROUTINE OUT
 WRITE (6,10)
 WRITE (6,20)
 WRITE (6,30)
 WRITE (6,40)
 WRITE (6,50)
 WRITE (6,60)
 WRITE (6,70)
 WRITE (6,80)
 WRITE (6,90)
 WRITE (6,100)
 WRITE (6,110)
 RETURN

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FORMAT (23H) DEFINITIONS OF SYMBOLS (//)
 1 AREA RATIO/96H DELREF = GEOMETRIC AREA RATIO/31H ARATEF = EFFECTIVE
 2 LAT-FACED MODEL, DIVIDED BY MODEL NOSE RADIUS/99H DELREH = EQUILIBRIUM SHOCK STANDOFF DISTANCE ON FOUT
 3 BARIUM SHOCK STANDOFF DISTANCE ON HEMISPHERICAL MODEL, DIVIDED BY MOUT
 4 MODEL NOSE RADIUS/91H DELRFF = FROZEN SHOCK STANDOFF DISTANCE ON FOUT
 5 LAT-FACED MODEL, DIVIDED BY MODEL NOSE RADIUS/94H DELRFH = FROZEN
 6 SHOCK STANDOFF DISTANCE ON HEMISPHERICAL MODEL, DIVIDED BY MODEL
 7 NOSE RADIUS/73H DELSTR = DISPLACEMENT THICKNESS OF BOUNDARY LAYER
 8 ON NOZZLE WALL (INCH)
 9 FORMAT (64H) DELSTW = BOUNDARY LAYER DISPLACEMENT THICKNESS ON WEDOUT
 10 GE (INCH)/35H DIAM = NOZZLE DIAMETER (INCHES)/56H EPSLE = DENOUT
 11 SITY RATIO ACROSS EQUILIBRIUM NORMAL SHOCK/51H EPSLF = DENSITY RATIO
 12 3ATIO ACROSS FROZEN NORMAL SHOCK/52H GAMMA = FROZEN SPECIFIC-HEAT RATIO
 13 4 RATIO IN FREE STREAM/43H H = ENTHALPY IN FREE STREAM (BTU/LOUT
 14 5H)/58H HEIGHT = SECOND TRANSVERSE DIMENSION OF CHANNEL (INCHES)/30UT
 15 67H HR = RECOVERY ENTHALPY (BTU/LB)/74H HRATE = RATIO OF DISOUT
 16 7 SOCIATION ENTHALPY TO (HS-HW) FOR EQUILIBRIUM SHOCK)
 17 8 FORMAT (69H) HRATEF = RATIO OF DISSOCIATION ENTHALPY TO (HS-HW) FOUT
 19 1R FROZEN SHOCK/39H HS = STAGNATION ENTHALPY (BTU/LB)/84H K2PFOUT
 20 2E = SHOCK-BOUNDARY LAYER INTERACTION PARAMETER PER FOOT FOR EQUIOUT
 21 3LIBRIUM SHOCK/79H K2PFF = SHOCK-BOUNDARY LAYER INTERACTION PARAMOUT
 22 4ETER PER FOOT FOR FROZEN SHOCK/83H LEE = ATOM-MOLECULE LEWIS NOUT
 23 5UMBER AT STAGNATION CONDITION FOR EQUILIBRIUM SHOCK/78H LEF = OUT
 24 6ATM-MOLECULE LEWIS NUMBER AT STAGNATION CONDITION FOR FROZEN SHOCOUT
 25 7K/37H M = MACH NUMBER IN FREE STREAM/47H MU = VISCOSITYOUT
 26 8 IN FREE STREAM (LB/FT-SEC)/78H MUT2E = VISCOSITY AT STAGNATION OUT
 27 9CONDITION FOR EQUILIBRIUM SHOCK (LB/FT-SEC)/73H MUT2F = VISCOSITOUT
 28 0Y AT STAGNATION CONDITION FOR FROZEN SHOCK (LB/FT-SEC)/52H MW =
 29 \$ = MOLECULAR WEIGHT IN FREE STREAM (GM/MOLE)
 30 1 FORMAT (83H) MW2E = MOLECULAR WEIGHT AT STAGNATION CONDITION FOROUT
 31 2 EQUILIBRIUM SHOCK (GM/MOLE)/78H MW2F = MOLECULAR WEIGHT AT STADOUT
 32 3GNATION CONDITION FOR FROZEN SHOCK (GM/MOLE)/53H N = BOUNDAROUT
 33 4Y-LAYER PRESSURE-GRAIDENT PARAMETER/40H P = PRESSURE IN FREEOUT
 34 5 STREAM (ATM)/71H PRE = PRANDTL NUMBER AT STAGNATION CONDITIONOUT
 35 6 FOR EQUILIBRIUM SHOCK/66H PRF = PRANDTL NUMBER AT STAGNATION OUT
 36 7CONDITION FOR FROZEN SHOCK/68H PRREF = PRANDTL NUMBER AT REFERENOUT
 37 8CE TEMPERATURE IN BOUNDARY LAYER/58H PT2E = STAGNATION PRESSUREOUT
 39 9 FOR EQUILIBRIUM SHOCK (ATM)/53H PT2F = STAGNATION PRESSURE FOROUT

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111 = AXIAL DOUT
 112 DOWNSTREAM DOUT
 113 POINTS
 114 OF WEDGE, OUT
 115 = ORDINATE OF
 116 PASSING THROUGH LEAD
 117 = NONDIMENSIONAL STREAMWISE
 118 DOUT
 119-

FIRST TRANSVERSE DIMENSION OF CHANNEL (INCHES)/93H X
 71 DISTANCE ALONG NOZZLE, MEASURED FROM THROAT AND POSITIVE
 8 (INCHES)/67H XSN = N AVERAGED OVER SEVERAL PRECEDING
 9 OF THE SOLUTION/85H XW = DISTANCE FROM LEADING EDGE OF WEDGE,
 \$ MEASURED ALONG WEDGE SURFACE (INCHES)/115H YS = ORDINATE OF
 \$ SHOCK FROM LINE PARALLEL TO FREE STREAM FLOW, PASSING THROUGH LEAD
 \$ COORDINATE OF WEDGE (INCHES)/71H ZETA = NONDIMENSIONAL STREAMWISE
 \$ COORDINATE IN BLUNT WEDGE ANALYSIS
 END

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50  FAREA=2.*RTHCM(1)
60  GO TO 100
70  IF (NPRFLS.EQ.1) GO TO 80
80  DO 70 I=1,2
90  DELSTP(I)=(1.-OMDST(I))*RO
100 FAREA=4.*(RTHCM(1)-DELSTP(1))*(RTHCM(2)-DELSTP(2))
110 GO TO 100
120 IF (JDIM.EQ.0) GO TO 90
130 FAREA=PZ*(OMDST(1)*RTHCM(1))*#2
140 GO TO 100
150 FAREA=FAREA/929.03
160 WRITE (6,570)
170 TFLOW=FAREA*CF(1)*CF(2)*SM
180 IF (ISW2B.GT.0) GO TO 110
190 WRITE (6,580) FLOW
200 IF (IC.FQ.0) GO TO 120
210 DEL=SAVEC(1)*RHO*TB7/CM
220 GO TO 130
230 DEL=0.
240 IF (NPRFLS.EQ.2) GO TO 140
250 WRITE (6,590) FACNAM,DIAM(1)
260 GO TO 150
270 WRITE (6,600) CHANAM,DIAM(1),DIAM(2),FACNAM
280 WRITE (6,610) POUT,TOUT,HOUT,SOUT,RCUT,UCUT,FLX,TFLOW,GAMMA,CM,DEL
290 *WRITE (6,620)
300 WRITE (6,630) (HP(I),SAVEC(I),I=1,ISS)
310 RETURN
320 FVOUT(1)=CF(4)*CX
330 FVOUT(2)=CT*CTAP
340 FVOUT(3)=CF(5)*CH
350 FVOUT(4)=PRES*PRESA
360 FVOUT(5)=CF(1)*RHO
370 FVOUT(7)=CF(2)*SU
380 FVOUT(8)=AMACH
390 FVOUT(9)=SEN
400 FVOUT(11)=AFNX
410 I6=I6+1
420 IF (ISW3B.EQ.0) GO TO 180
430 DO 170 I=1,NPRFLS
440 I1=7+10*I
450 FVOUT(I1)=CF(6)*DELBL(I)
460 FVOUT(I1+1)=CF(6)*THETA(I)
470 FVOUT(I1+2)=QWDOT(I)
480 FVOUT(I1+3)=TAUW(I)
490 FVOUT(I1+4)=1.8*HR
500 FVOUT(I1+5)=PRREF
510 CONTINUE
520 VARNAM(6)=DNAM
530 VARNAM(15)=ANAM
540 IF (NPRFLS.EQ.1) GO TO 200
550 CALL GVAR2 (CX,Y,Z)
560 IF (ERR) RETURN
570 IF (ISW3B.NE.0) GO TO 190
580 CHDIM(2)=2.*Y*CF(4)

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190 CHDIM(1)=2.*Z*CF(4)
    FVOUT(6)=1.E20
    GO TO 200
    FVOUT(16)=2.*Z*CF(4)
    FVOUT(26)=2.*Y*CF(4)
    VARNAM(16)=WNAME
    CONTINUE
200 FVOUT(10)=GAMMA
    IF (ISW3B.NE.0) GO TO 250
    IF (NPRFLS.EQ.2) GO TO 210
    CALL GMAR (CX,Y)
    IF (ERR) RETURN
    FVOUT(6)=2.*Y*CF(4)
    IF (NOTRAN) GO TO 220
    CALL TRANSP (FVOUT(2),FVOUT(4))
    IF (ERR) RETURN
    FVOUT(14)=VISC*CF(3)
    IF (VISC.EQ.0) GO TO 230
    FVOUT(12)=FVOUT(5)*FVOUT(7)/FVOUT(14)
    GO TO 240
    FVOUT(12)=0.
    FVOUT(15)=SIGMA
    GO TO 260
    FVOUT(12)=REPF
    FVOUT(14)=CF(3)*VISC.
    FVOUT(15)=SIGMA
    FVOUT(13)=CM
    HJ=1.8*CHA*CRP/CMA
    IF (ISW3B.EQ.0) GO TO 340
    IF (FVOUT(7).EQ.0.) GO TO 270
    STFAC=1./(FVOUT(5)*FVOUT(7))*(H0-HW)
    FVOUT(23)=STFAC*FVOUT(19)
    FVOUT(24)=FVOUT(12)*FVOUT(18)/12.
    FVOUT(25)=200.*EXP(0.224*FVOUT(8))
    IF (NPRFLS.EQ.1) GO TO 280
    FVOUT(33)=STFAC*FVOUT(29)
    FVOUT(34)=FVOUT(12)*FVOUT(28)/12.
    FVOUT(35)=FVOUT(25)
    IF (LIMOUT.EQ.0) GO TO 300
    DO 290 I=1,NPRFLS
    FVOUT(1,I)=SN(I)
    FVOUT(2,I)=XSN(I)
    CALL GEMAR (CX,ARAT,DARAT)
    IF (ERR) RETURN
    IF (NPRFLS.EQ.2) GO TO 310
    FVOUT(16)=ARAT
    GO TO 320
    FVOUT(6)=ARAT
    VARNAM(6)=ANAM
    GO TO 340
    IF (JDIM.EQ.0) GO TO 330
    FVOUT(6)=RTHCM(1)*SQRT(ARAT)*CF(4)*2.
    GO TO 340
    FVOUT(6)=RTHCM(1)*ARAT*CF(4)*2.
    INEQI=INEQ+1

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WRITE (6,640) (ASTRSK,I=1,8),(ASOLN(J,1,CLN),(ASTRSK,K=1,3),J=1,3),(ASTRSK,K=1,3)
1,NSTEPS,STEP(ISCLN),(ASTRSK,L=1,4),TYPISLN(INEOP1,ISOLN)
WRITE (6,650) (VARNAM(I),FVOUT(I),I=1,15)
IF (ISW3B.EQ.0) GO TO 360
DO 350 L=1,NPRFLS
I1=6+10*L
I2=I1+9
WRITE (6,660) (VARNAM(I),FVOUT(I),I=1,12)
CONTINUE
IF (NPRFLS.EQ.1,CR,ISW3B.NE.0) GO TO 370
WRITE (6,670) CHDIMS(1),CHDIMS(2)
IF (LIMOUT.EQ.0) GO TO 380
WRITE (6,680) ((VARN2(I),J,FV02(I,J),I=1,2),J=1,NPRFLS)
IF (NT.EQ.1,CR,ISOLN.NE.3) GO TO 390
TNDOUT(1)=CTAP*CTE
TNDOUT(2)=ODPR
TNDOUT(3)=ODPE
TNDOUT(4)=H0
WRITE (6,690) (TNNAM(I),TNDOUT(I),I=1,4)
IF (ISW68.EQ.0,CR,ISOLN.EQ.1) GO TO 440
IF (I6.LT.IS6) GO TO 440
I6=0
WRITE (6,690)
IF (IC.EQ.0) GO TO 400
EMF=SAVEC(I)
SAVEC(I)=EMF*RHO*TB7/CM
ISSSP1=(ISS-1)/5+1
DO 410 K=1,ISSSP1
LIM1=5*(K-1)+1
LIM2=MINO(5*K,ISS)
WRITE (6,700) (HD(I),SAVEC(I),I=LIM1,LIM2)
IF (IC.EQ.0) GO TO 420
SAVEC(I)=EMF
IF (ISW58.GT.0,CR,ISOLN.NE.3) GO TO 440
WRITE (6,510)
WRITE (6,520) (I,PI(I),I=1,ISR)
WRITE (6,530) (I,CHI(I),I=1,ISR)
WRITE (6,540) (I,PICHI(I),I=1,ISR)
DO 430 I=1,ISS
DLGDX(I)=DGJ(I)/GJ(I)
WRITE (6,550) (HP(I),DLGDX(I),I=1,ISS)
IF (.NOT.DATAPE) GO TO 500
IF (ITYPER.NE.1) GO TO 450
IMP=0
IX=0
IX=IX+1
NRCOUT=NRCOUT+1
ITYPER=2
XXX(1)=FVOUT(1)
XXX(2)=FVOUT(2)
XXX(3)=FVOUT(4)
XXX(4)=FVOUT(5)
XXX(5)=FVOUT(7)
XXX(6)=FVOUT(3)
XXX(7)=FVOUT(8)

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460 IF (ISW3B) 460,480,460
    IBL=17+10*(NBL-1)
    XXX(9)=FVOUT(1BL)
    IF (NPRELS.EQ.2) GO TO 470
    XXX(8)=FVOUT(16)
    GO TO 490
    XXX(8)=FVOUT(6)
    GO TO 490
480 XXX(8)=FVOUT(11)
    XXX(9)=0.
490 XXX(10)=FVOUT(12)
    WRITE (ITPUT) ITPER,IX,(XXX(I),I=1,10)
    RETURN
C
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510 FORMAT (57X,18HREACTION RATE DATA)
520 FORMAT (3H P.,11X,8(I4,1PE1C,1)/9(I4,1PE10,1),6X)
530 FORMAT (4H CHI,10X,8(I4,1PE10,1)/9(I4,1PE10,1),6X)
540 FORMAT (6H PICH1,1X,8(I4,1PE1C,1)/9(I4,1PE10,1),6X)
550 FORMAT (4H DLG,12X,7(2X,A4,1PE10,1)/8(2X,A4,1PE10,1),4X)
560 FORMAT (1X,A3,1H(.11,3H) =.4X,1PE10.3,4X,3(A3,1H(.11,3H) =.4X,E.0.
    1J,4X))
570 FORMAT (1H0)
580 FORMAT (19H GAS FLOW RATE =,F10,3,2X,7H LB/SEC)
590 FORMAT (10H NOZZLE - ,A4,5X,F10,3,2X,21H INCH THROAT DIAMETER)
600 FORMAT (9H CHANNEL ,A4,6X,F10,3,3H BY,F7,3,17H INCH THROAT FOR ,A4
    1)
610 FORMAT (19H PRESSURE =,F1,3,2X,4H ATM/19H TEMPERATURE
    1 =,F7,0,5X,6H DEG K/19H ENTHALPY =,F7,0,5X,7H BTU/LB',9H
    2ENTROPY =,F9,2,3X,13H BTU/LB-DEG R/19H DENSITY =,F7,0,5X,7H FT/SEC/19H
    3,FI2,S,9H LB/CU FT/19H VELOCITY =,F7,0,5X,7H FT/SEC/19H MAOUI
    4SS FLUX =,F10,3,2X,13H LB/SQ FT-SEC/19H COMPUTED FLOW =,0,0,1
    5F10,3,2,7H LB/SEC/19H GAMMA =,F10,3/19H MOLECULAR WEIGHT
    6HT =,F9,2,3X,8H GM/MOLE/19H ELECTRON DENSITY =.3X,1PE9.2,13H FLECT
    7RONS/CC)
620 FORMAT (/10X,22HSPECIES MOLE FRACTIONS/)
630 FORMAT (13X,A4,4X,1PE10,3)
640 FORMAT (/76H ***,14A6,16,6A6//)
650 FORMAT (1X,A6,2H =,F11,3,7X,A6,2H =,F8,0,10X,A6,2H =,F8,0,10X,A6,2H
    1H =.4X,1PE10,3,4X,A6,2H =,4X,E10,3/1X,A6,2H =,0PF11,3,7X,A6,2H =,F0U1
    28,0,10X,A6,2H =,F11,3,7X,A6,2H =,F10,2,8X,A6,2H =,F11,3/1X,A6,2H =,0
    3,F11,3,7X,A6,2H =,4X,1PE10,3,4X,A6,2H =,0PF10,2,8X,A6,2H =,4X,1PE1
    4,3,4X,A6,2H =,4X,F10,3)
660 FORMAT (1X,A6,2H =,F11,3,7X,A6,2H =,F11,3,7X,A6,2H =,F11,3,7X,A6,2H
    1H =.4X,1PE10,3,4X,A6,2H =,4X,E10,3/1X,A6,2H =,0PF8,0,10X,A6,2H =,F0U1
    212,4,6X,A6,2H =,4X,1PE10,3,4X,A6,2H =,0PF8,0,10X,A6,2H =,F8,0)
670 FORMAT (9X WIDTH =,F11,3,6X,9H HEIGHT =,F11,3)
680 FORMAT (1X,A6,2H =,F8,0,10X,A6,2H =,4X,1PE10,3,4X,A6,2H =,4X,10,3
    1,4X,A6,2H =,4X,F10,3)
690 FORMAT (45X,41HSPECIES MOLE FRACTIONS IN THE FREE STREAM)
700 FORMAT (5(1X,14,3X,1H=.4X,1PE10,3,3X))
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SUBROUTINE PERT
LOGICAL ERR
REAL ACOM(30),ELEMENT(10),HP(20)
DOUBLE PRECISION AA,AAA,CAPX,GJ,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON C, CARB, CRB, CSTA, CTT, DLOGA, PRESA, RHOP, SHPG, SC, TSTOP,
1 CRRB, CRB, CSTA, CTT, DLOGA, PRESA, RHOP, SHPG, SC, TSTOP,
2 CTP, CTPL, DELTAX, DLOGA, PRESA, RHOP, SHPG, SC, TSTOP,
3 DELT2, DELTAX, DLOGA, PRESA, RHOP, SHPG, SC, TSTOP,
4 SPCTEST, PRESA, RHOP, SHPG, SC, TSTOP,
5 GRMDBAR, RMOC, RHOP, SHPG, SC, TSTOP,
6 TEST, TESTB, TPRINT, TSTOP,
COMMON BE(64), CACT(64), CLNPT(64), PERIGJ(20), SAJ(20), SS(20), TFE(20),
1 CCPJ(20), CACT(64), CLNPT(64), PERIGJ(20), SAJ(20), SS(20), TFE(20),
2 CLNMC(64), CLNPT(64), PERIGJ(20), SAJ(20), SS(20), TFE(20),
3 GUB(64), PERIGJ(20), SAJ(20), SS(20), TFE(20),
4 GO(64), SAJ(20), SS(20), TFE(20),
5 SKIL(20), SAJ(20), SS(20), TFE(20),
6 TFD(20), SAJ(20), SS(20), TFE(20),
COMMON BETA(64,20), XNUIJP(64,20),
1 XNUIJ(64,20), XNUIJP(64,20),
COMMON IC, ISR, ISW2A, ISW6A, ISW6B, NNS, IGM(20),
1 ISMC, ISVNR, ISR, ISW2A, ISW6A, ISW6B, NNS, IGM(20),
2 ISW1A, ISW1B, ISW2A, ISW2B, IUPD, IZER, NQT, NTEST,
3 ISW5B, ISW6A, ISW6B, IUPD, IZER, NQT, NTEST,
4 NIT, NNN, NNS, NQS,
COMMON IGJ(20),
COMMON ACOM, ELEMENT, HP,
DIMENS:ON AAA(22,24),RTA(64,20),THEVP(20)
1J(20),SDGJ(20),SHJAP(20),
EQUIVALENCE (AA(1,1),AAA(1,1)),(BETA(1,1),RTA(1,1)),
1 (BLBK(1),CAPO(1)),(CAPXTH(1),CCI(1)),(GJA(1),DGJ(1)),
2 (APX(1),GJ(1)),(SAJ(1),SBJ(1)),(SS(1),SDGJ(1)),
3 (JA(1),SHJAP(1)),(THEV(1),THEVP(1))
COMMON /ERROR/ ERR
DO 10 I=1,ISSP2
DO 10 J=1,ISSP3
AA(I,J)=0
DO 20 I=1,ISS
DO 20 J=1,ISS
AA(I,J)=LPIJ(J,I)
DO 40 I=ISSP1,ISS
DO 40 J=1,ISS
Z=0
DO 30 K=1,ISR
Z=Z+(BETA(K,I)*PI(K)*BETA(K,J))
Z=Z/GJ(J)
AA(I,J)=-Z
DO 50 J=1,ISS
AA(I,ISSP1:J)=(CMA*SHJ(J)/(SU2))
Z1=0
DO 70 K=1,ISR

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60 Z=0.
70 DO 60 J=1,ISS
80 Z=Z+(BETA(K,J))*((SHJ(J)/CT)-1.)
90 Z1=Z1+Z*(BETA(K,I)*PI(K))
100 AA(I,ISSP1)=Z1/CT
110 Z=0.
120 DO 90 J=1,ISS
130 Z=Z+GJ(J)*CCPJ(J)
140 AA(I,ISSP1,ISSP2)=(CMA*Z)/(SU2)
150 Z1=0.
160 DO 100 K=1,ISR
170 Z1=Z1+(BETA(K,I))*PI(K)*BE(K)
180 AA(I,ISSP2)=-Z1/RHO
190 AA(I,ISSP1,ISSP2)=-1./RHO
200 DO 120 I=1,ISS
210 AA(I,ISSP2,I)=SENT(I)-1.0-ALOG(GJ(I))*RHO*CT*82.055*RHAP*CTAP)
220 Z=0.
230 DO 130 J=1,ISS
240 Z=Z+GJ(J)*((CCPJ(J))-1.)
250 AA(I,ISSP2,ISSP1)=Z/CT
260 AA(I,ISSP2,ISSP2)=-1./(CM*RHO)
270 DO 140 I=1,ISS
280 AA(I,ISSP3)=DGJ(I)
290 CALL DSMSQL(AA,ISSP2,22)
300 IF (ERR) RETURN
310 DO 150 I=1,ISS
320 PERTGJ(I)=AA(I,ISSP3)
330 PCT=AA(I,ISSP1,ISSP3)
340 PRHO=AA(I,ISSP2,ISSP3)
350 DO 170 I=1,ISS
360 Z=0.
370 DO 160 J=1,ISS
380 Z=Z+BETA(I,J)*((SHJ(J)/CT)-1.)
390 Z1=Z1+(BETA(I,J))*PERTGJ(J)/GJ(J)
400 SOCHI(I)=-BE(I)*PRHO/RHO+Z*PCT/CT-Z1
410 RETURN
420 END

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FUNCTION PIOMEG(TE)
COMMON /TNEG/ TLIST(30), POM(30)
DO 10 I=2,30
IF (TE.LT.TLIST(I)) GO TO 20
CONTINUE
PIOMEG=POM(30)
GO TO 30
PIOMEG=POM(I-1)+(POM(I)-POM(I-1))*(TE-TLIST(I-1))/(TLIST(I)-TLIST(I-1))
RETURN
END

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SUBROUTINE PRTA
LOGICAL ERR,MODLPT
REAL ACOM(30),ELEMENT(10),HP(20)
DOUBLE PRECISION A,CAPX(20)
COMMON AA(22,24),L,IJ(20,10),CAPX(20)
COMMON A, AFNTS, AFNX, AMACH,
C, CARL, CRS, CSTA, CH,
CTP, CTPL, CTT, CTAP, CLNT, CLTAP,
DELTA2, DELTAY, DLOGA, DLOGR, DLOGR, DT,
SPCTEST, PRES, PRESB, PRESTH, PRHO, RHAP, RHO,
GRHOBAR, RHOC, RHOD, RHO, RHTH, ROBAR, ROBARP,
SEN, SHPG, SC, SL, SL64,
TEST, TESTB, TPRINT, TSTOP, UP,
COMMON BE(64), BLBK(31), CAI(64), CAPXTH(20),
CCPJ(20), CFACT(64), CGI(20), CHI(64),
2CLNIMC(64), CLNPI(64), CMW(20), ETAI(64),
GJB(20), PERTGJ(20), PGJ(20), PI(64), PCHI(64),
4 SKIL(20), SAJ(20), SDCHI(64), SENT(20), SHJ(20),
TFD(20), TFE(20), TFK(20), TB(30), TFA(20),
COMMON BETA(64,20), ELJ(10,20), GELJ(10,20),
XNUIJ(64,20),XNUIJP(64,20)
COMMON IC, INEQ, INEQV, IRUN, ISC, ISCP1,
ISMC, ISMCNR, ISR, ISS, ISSNR, ISSP1, ISSP2, ISSP3, ISSP4,
2 ISW1A, ISW1B, ISW2A, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
3 ISW5B, ISW6A, ISW6B, IUPD, IZERO, IZJK, LC, MI, NFIT,
NIT, NNN, NNS, NQS, NGT, NTEST
COMMON IGJ(20), IGM(20), ITB(5), KUR(64,20), LPIJ(20,10)
COMMON ACOM, ELEMENT, HP
COMMON /BL/ DELBL(2),BLINT(2),XZERO,TWALL,CPWALL,VISROT,DIAM(2),
COMMON /NEO/ OMDST(2),DDELBL(2)
COMMON /SS/ CAS,US
COMMON /ERROR/ ERR
COMMON /MODP2/ MODLPT
COMMON /STEPS/ NSTEPS
COMMON /TONEQ/ CTF,DTE,BPAR,EPAR(2,25),NT,ITR(25),KTF(25),
1 KTR(25),IC(20),IPA(25)
COMMON /PRTDMP/ AFNTS, CAPX,CHI,DLOGA,DLOGR,DT,IUPD,
1 PCT,PERTGJ,PI,FRHO,SDCHI,CAS,DELBL,OMDST,DDELBL
NSTEPS=NSTEPS+1
IF (ISW3R.EQ.0) GO TO 10
AS=SU*US/AMACH
CASE=AS*AS/CT
CALL BLCALL (.TRUE.)
IF (ERR) RETURN
IF (ISW5B.NE.0) WRITE (6,PRTDMP)
IF (INEQ.NE.0) GO TO 20
CT=CT+PCT
CTE=CT
RHO=RHOC+PRHO
IF (ABS(CT-CTT).LT.TPRINT.AND..NOT.MODLPT) GO TO 30
CTT=CT
CALL OUT2 (3)

```

```

1 PRT
2 PRT
3 PRT
4 PRT
5 PRT
6 PRT
7 PRT
8 PRT
9 PRT
10 PRT
11 PRT
12 PRT
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41 PRT
42 PRT
43 PRT
44 PRT
45 PRT
46 PRT
47 PRT
48 PRT
49 PRT
50 PRT
51 PRT
52 PRT
53 PRT
54 PRT
55 PRT

```

```
IF (ERR) RETURN
NSTEPS=0
IF (.NOT. MODLPT) GO TO 30
CALL MODEL
MODLPT=.FALSE.
IF (INEQ.NE.0) GO TO 40
CT=CT-PCT
CTE=CT
RHO=RH0-PRHO
RETURN
END
```

30

40
C

```
PRT 56
PRT 57
PRT 58
PRT 59
PRT 60
PRT 61
PRT 62
PRT 63
PRT 64
PRT 65
PRT 66
PRT 67-
```

SUBROUTINE PUTGIN (X)
 SUBROUTINE PUTGIN SUPERVISES THE COMPUTATION OF COLLISION CROSS
 SECTIONS FOR THE TRANSPORT CALCULATIONS.

LOGICAL WRITE
 DIMENSION OM(3)
 COMMON /TRANS1/ T,Q(3,20,20),ZM2(20)
 COMMON /TRANS7/ V(400),KQ(100),NQ(100),IQ(400),JQ(400),NKQ
 COMMON /TRANS8/ NV(15),N
 COMMON /RDTR/ ISW8B

DATA IDBG /-1/
 IDBG=IDBG+1
 WRITE=,FALSE.

INITIALIZE Q ARRAY TO ZERO.
 DO 10 I=1,N
 DO 10 J=1,N
 DO 10 K=1,3

O(K,I,J)=0.
 CALL QCOUL (OM,X,V)
 O(1,2,1)=OM(1)
 COMPUTE CROSS SECTION VALUES AND ACCUMULATE IN Q ARRAY.
 M=2

LQ1=1
 DO 240 L=1,NKQ
 K=KQ(L)
 LQ2=NQ(L)

GO TO (20,30,40,50,60,70,80,90,100,110,120,130,140,150,160), K
 CONTINUE

GO TO 190
 OM(1)=V(M)*Q(1,2,1)
 OM(3)=OM(1)*1.5625
 OM(2)=V(M+1)*OM(1)

GO TO 190
 CALL QEXP (OM,V(M))
 GO TO 190

CALL QEX (OM,V(M))
 IF (V(M+3)) 190,190,170
 CALL QTAB (OM,V(M))

GO TO 190
 CALL GREPP (OM,V(M))
 GO TO 190

CONTINUE
 17 CALL QATTP (OM,V(M))
 GO TO 190

CALL QLJ (OM,V(M))
 GO TO 190
 CALL QSAME (JM,V(M))

GO TO 190
 CALL QMIX (LQ1,LQ2)
 GO TO 210

CALL Q11 (LQ1,LQ2,V(M))
 GO TO 210
 CALL Q12 (OM,V(M))
 GO TO 190

CALL Q13 (LQ1,LQ2,V(M))
 GO TO 210

1 PUT
 2 PUT
 3 PUT
 4 PUT
 5 PUT
 6 PUT
 7 PUT
 8 PUT
 9 PUT
 10 PUT
 11 PUT
 12 PUT
 13 PUT
 14 PUT
 15 PUT
 16 PUT
 17 PUT
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 47 PUT
 48 PUT
 49 PUT
 50 PUT
 51 PUT
 52 PUT
 53 PUT
 54 PUT
 55 PUT

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15| CALL Q14 (L01,L02,V(M))
16| GO TO 210
170| CONTINUE
180| GO TO 210
190| DO 180 LQ=L01,L02
200| I=IQ(LQ)
210| J=JQ(LQ)
220| O(1,I,J)=OM(1)
230| O(3,I,J)=OM(3)
240| GO TO 210
250| DO 200 LQ=L01,L02
260| I=IQ(LQ)
270| J=JQ(LQ)
280| DO 200 KK=1,3
290| O(KK,I,J)=O(KK,I,J)+OM(KK)
300| M=M+NV(K)
310| C
320| DEBUG OUTPUT
330| IF (ISW8.EQ.0) GO TO 240
340| IF (IDBG.EQ.0) GO TO 220
350| IF (IDBG.NE.ISW8) GO TO 240
360| WRITE=.TRUE.
370| WRITE (6,250) L,K,NO(L),M,LQ1
380| DO 230 I=1,N
390| WRITE (6,260) (O(1,J,I),J=1,I)
400| LQ1=LQ2+1
410| IF (WRQTE) IDBG=0
420| RETURN
430| C
440| FORMAT (1HC/10IS/)
450| FORMAT (1P10E12.4)
460| END

```

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56| PUT
57| PUT
58| PUT
59| PUT
60| PUT
61| PUT
62| PUT
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64| PUT
65| PUT
66| PUT
67| PUT
68| PUT
69| PUT
70| PUT
71| PUT
72| PUT
73| PUT
74| PUT
75| PUT
76| PUT
77| PUT
78| PUT
79| PUT
80| PUT
81| PUT
82| PUT
83| PUT
84| PUT
85| PUT
86|-

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```

SUBROUTINE QCOUL (OM1,X,C)
SUBROUTINE QCOUL COMPUTES THE FACTOR 0.8*QC IN THE EFFECTIVE
COULOMB CROSS SECTION.
COMMON /TRANS1/ T,Q(3,20,20),ZM2(20)
OM1=2.2346E10/T**2
IF (C) 20,20,10
Y=(T**4)/(1.E15*X+1.E-35*(T**4))
OM1=.5*OM1*ALOG(2.091E1*Y+1.517E2*Y**.666667)
RETURN
END

```

C
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QCO
QCO
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QCO
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QCO
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QCO
QCO

SUBROUTINE QEX (OM,V)
 SUBROUTINE QEX COMPUTES CROSS SECTIONS FOR A RESONANT EXCHANGE
 PROCESS.
 DIMENSION OM(3),V(3)
 COMMON /TRANSI/ T,O(3,20,20),ZM2(20)
 A=V(1)
 B=V(2)
 ZM=V(3)
 Y=A-.217147*B#ALOG(T/ZM)
 OM(1)=2.*(Y-4.461*B)**2+.03724*B**2
 OM(3)=2.*(Y-4.317*B)**2-.06754*B**2
 OM(2)=0.
 RETURN
 END

1 QEX
 2 QEX
 3 QEX
 4 QEX
 5 QEX
 6 QEX
 7 QEX
 8 QEX
 9 QEX
 10 QEX
 11 QEX
 12 QEX
 13 QEX
 14- QEX

C

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SUBROUTINE QINTRP (A,B,N1,N)
 SUBROUTINE QINTRP INTERPOLATES TABULAR DATA FOR THE CROSS SECTION
 COMPUTATIONS

```

COMMON /RDTR/ ISW8B
DIMENSION B(3)
COMMON /TRANS4/ TL(1000),OM(1000,3)
I=N1
IF (TL(I)-A) 10,40,30
N2=N1+N-2
DO 20 I=N1,N2
IF (TL(I+1)-A) 20,40,40
CONTINUE
I=N2
IF (ISW8B.EQ.0) GO TO 40
WRITE (6,60) I,A,TL(I),TL(I+1)
DO 50 J=1,3
B(J)=OM(I+1,J)+(A-TL(I+1))*(OM(I,J)-OM(I+1,J))/(TL(I)-TL(I+1))
RETURN
FORMAT (25H0 EXTRAPOLATED VALUES,1P4E14.4/1H )
END
  
```

1 QIN
 2 QIN
 3 QIN
 4 QIN
 5 QIN
 6 QIN
 7 QIN
 8 QIN
 9 QIN
 10 QIN
 11 QIN
 12 QIN
 13 QIN
 14 QIN
 15 QIN
 16 QIN
 17 QIN
 18 QIN
 19 QIN
 20 QIN
 21-

C

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C 60


```

SUBROUTINE QLJ (OM,V)
SUBROUTINE QLJ COMPUTES CROSS SECTIONS FOR THE LENNARD-JONES 6-12
POTENTIAL.
DIMENSION OM(3), V(3)
COMMON /TRANS1/ T,0(3,20,20),ZM2(20)
EPSLN=V(1)
SIGMA=V(2)
NI=V(3)
TSTAR=T/EP SLN
CALL QINTRP (TSTAR,OM,N1,37)
OM(1)=3.1416*SIGMA**2*OM(1)
OM(2)=OM(2)*OM(1)
OM(3)=OM(3)*CM(1)
RETURN
END

```

```

1 QLJ
2 QLJ
3 QLJ
4 QLJ
5 QLJ
6 QLJ
7 QLJ
8 QLJ
9 QLJ
10 QLJ
11 QLJ
12 QLJ
13 QLJ
14 QLJ
15- QLJ

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C
C
SUBROUTINE QMIX (LQ1,LQ2)
SUBROUTINE QMIX COMPUTES CROSS SECTIONS FROM THE EMPIRICAL MIXING
RULE
DIMENSION SQT(3,20)
COMMON /TRANS1/ T,Q(3,20,20),ZM2(20)
COMMON /TRANS7/ V(40),KQ(100),NG(100),IG(400),JQ(400),NKQ
IF (LQ2.LT.LQ1) GO TO 70
DO 10 I=1,20
SQT(I,I)=0.
DO 60 LQ=LQ1,LQ2
I=IQ(LQ)
J=JQ(LQ)
IF (SQT(I,I).GT.0.) GO TO 30
DO 20 K=1,3
SQT(K,I)=.5*SQT(Q(K,I,I))
IF (SQT(I,J).GT.0.) GO TO 50
DO 40 K=1,3
SQT(K,J)=.5*SQT(Q(K,J,J))
DO 60 K=1,3
Q(K,I,J)=(SQT(K,I)+SQT(K,J))*2+Q(K,I,J)
RETURN
END
10
20
30
40
50
60
70

```

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QMIX
QMIX
QMIX
QMIX
QMIX
QMIX
QMIX
QMIX
QMIX
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QMIX
QMIX
QMIX
QMIX

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SUBROUTINE GREPP (OM, VV)
SUBROUTINE GREPP COMPUTES CROSS SECTIONS FOR AN INVERSE POWER LAW
INTERACTION POTENTIAL.
DIMENSION OM(3), VV(3)
COMMON /TRANS1/ T, O(3, 20, 20), ZM2(20)
COMMON /TRANS4/ TL(1000), OMEGAI(1000), ASTAR(1000), BSTAR(1000)
ITL=VV(1)
OM(1)=OMEGAI(ITL)*T**(-2./VV(2))
OM(2)=FSTAR(ITL)*OM(1)
OM(3)=BSTAR(ITL)*OM(1)
RETURN
END

```

```

ORE
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ORE
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ORE
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ORE
ORE
ORE
ORE

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```

```

SUBROUTINE QSAME (OM,VV)
SUBROUTINE QSAME SETS THE CROSS SECTION FOR A GIVEN SPECIES PAIR
EQUAL TO THOSE FOR ANOTHER PAIR.
DIMENSION OM(3),VV(5)
JMMCN / TRANS1/ T,0(3,20,20),ZM2(20)
I=VV(1)
J=VV(2)
C=VV(3)
OM(1)=C*Q(1,I,J)
OM(2)=C*Q(2,I,J)*VV(4)
OM(3)=C*Q(3,I,J)*VV(5)
RETURN
END

```

```

1 QSA
2 QSA
3 QSA
4 QSA
5 QSA
6 QSA
7 QSA
8 QSA
9 QSA
10 QSA
11 QSA
12 QSA
13- QSA

```

C

SUBROUTINE QTAB (OM,V)
 USE TABULAR DATA FOR CROSS SECTIONS.
 DIMENSION CM(3),V(3)
 COMMON /TRANS1/ T,O(3,20),ZM2(20)
 AEV(1)
 N1=V(2)
 NLEV(3)
 CALL QINTRP (T,CM,N1,NL)
 CM(1)=A*CM(1)
 CM(2)=A*CM(2)
 CM(3)=OM(3)*OM(1)
 RETURN
 END

1
 2
 3
 4
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 11
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 13-
 QTA
 QTA
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SUBROUTINE Q11 (L01,L02,VI)
SUBROUTINE Q11 MULTIPLIES CROSS SECTIONS BY A RAMP FUNCTION OF
TEMPERATURE.
DIMENSION VV(3),VI(2)
COMMON /TRANS1/ T,0(3,20,20),ZM2(20)
DATA VV/3*1./
T0=VI(1)
T1=VI(2)
VV(1)=AMAX1(0.,(T-T0)/(T1-T0))
IF (VV(1).LT.1.) CALL Q14 (L01,L02,VV)
RETURN
END

```

```

1 011
2 011
3 011
4 011
5 011
6 011
7 011
8 011
9 011
10 011
11 011
12- 011

```

C
C

10

```

SUBROUTINE Q12 (QM,VV)
SUBROUTINE Q12 COMPUTES CROSS SECTIONS FROM THE GENERALIZED MIXING
RULE (EQ. 27),
DIMENSION QM(3),VV(4)
COMMON /TRANS1/ T,Q(3,20,20),ZM2(20)
I=VV(1)
J=VV(2)
K=VV(3)
L=VV(4)
DO 10 M=1,3
  QM(M)=.25*(SQRT(Q(M,I,J))+SQRT(Q(M,K,L)))**2
RETURN
END

```

```

1 2 3 4 5 6 7 8 9 10 11 12 13-
Q12 Q12 Q12 Q12 Q12 Q12 Q12 Q12 Q12 Q12 Q12 Q12 Q12

```

C

10

C

F

013	1
013	2
013	3
013	4
013	5
013	6
013	7
013	8
013	9
013	10
013	11
013	12
013	13
013	14-

```

SUBROUTINE Q13 (L01,L02,VV)
SUBROUTINE Q13 SETS TWO CROSS SECTIONS EQUAL FOR A SPECIES PAIR.
DIMENSION VV(3)
COMMON /TRANS1/ T,O(3,20,20),ZM2(20)
COMMON /TRANS7/ V(400),KG(100),NQ(100),IQ(400),JQ(400),NKG
K1=VV(1)
K2=VV(2)
C=VV(3)
DO 10 L0=L01,L02
I=IQ(L0)
J=JQ(L0)
Q(K1,I,J)=C*Q(K2,I,J)
RETURN
END

```

C

10

SUBROUTINE Q14 (LQ1,LQ2,VV)

SUBROUTINE Q14 MULTIPLIES CROSS SECTIONS BY A CONSTANT.

DIMENSION VV(3)

COMMON /TRANS1/ T,Q(3,20,20),ZM2(20)

COMMON /TRANS7/ V(400),KQ(100),NQ(100),IQ(400),JN(400),NKG

C2=VV(1)*VV(2)

C3=VV(1)*VV(3)

DO 10 LQ=LQ1,LQ2

I=IQ(LQ)

J=JO(LQ)

Q(1,I,J)=VV(1)*Q(1,I,J)

Q(2,I,J)=C2*Q(2,I,J)

Q(3,I,J)=C3*Q(3,I,J)

RETURN

END

1 Q14
2 Q14
3 Q14
4 Q14
5 Q14
6 Q14
7 Q14
8 Q14
9 Q14
10 Q14
11 Q14
12 Q14
13 Q14
14 Q14
15- Q14

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SUBROUTINE RADIUS ( ITYPE,X,R,AG,AGJ,L)
DIMENSION YZ(2),NPR(2)
COMMON /BL/ DELBL(2),BL INT(2),XZERO,TWALL,CPWALL,VISROT,DIAM(2),
1 SW,R0,JDIM,IPGINT
COMMON /BLRAD/ YOZO
COMMON /AREA/ ATPI(11,2),PARAM(3,12,2),RTHCM(2),NSECT(2),
1 NSECTU(2),ISHAPE(12,2),NPROFL(2),NPRFLS,NBL
DATA NPR /2,1/
GO TO (10,10,40), ITYPE
CALL GMAR (X,R)
RATIO=R/R0
GO TO (20,30,30), ITYPE
AG=RATIO
AGJ=1.
GO TO 50
AG=RATIO**2
AGJ=AG
GO TO 50
CALL GMAR2 (X,YZ(1),YZ(2))
AG=YZ(1)*YZ(2)/YOZO
R=YZ(L)
M=NPR(L)
AGJ=YZ(M)**2
RETURN
END

```

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1 RAD
2 RAD
3 RAD
4 RAD
5 RAD
6 RAD
7 RAD
8 RAD
9 RAD
10 RAD
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14 RAD
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16 RAD
17 RAD
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166 REA
167 REA
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177 REA
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179 REA
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181 REA
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218 REA
220 REA

DIMENSION EPRP(2,10)
 DIMENSION ELNAME(3),NATOM(3)
 DIMENSION EPRP(2,10),SPRP(43,30),RPRP(29,92),GPRP(124,6),
 ZPRP(64,10)
 DIMENSION NSECTC(2)
 DIMENSION SYI(3),SY2(3)
 DIMENSION TNEP(186,2)
 DIMENSION WORD(3,2)
 DIMENSION WORD2(2)
 EQUIVALENCE (AA(1,1),AAA(1,1)),(BETA(1,1),BTA(1,1)),
 (BLBK(1),CAPQ(1)),(CAPXTH(1),CCI(1)),(GJA(1),DGJ(1)),
 (CAPX(1),GJ(1)),(SAJ(1),SBJ(1)),(SS(1),SDGJ(1)),
 (SHJA(1),SHJAP(1)),(THEV(1),THEVP(1))
 EQUIVALENCE (CPI(1),CP(1,1))
 EQUIVALENCE (DUM(1),SUMGH)
 EQUIVALENCE (EPI(1),EPRP(1,1))
 EQUIVALENCE (EPI(1),EPRP(1,1))
 EQUIVALENCE (GPI(1),GPRP(1,1))
 EQUIVALENCE (NOZZLE,NPROFL(1))
 EQUIVALENCE (RPI(1),RPRP(1,1))
 EQUIVALENCE (SPI(1),SPRP(1,1))
 EQUIVALENCE (TNEP(1,1),TNI(1))
 EQUIVALENCE (ZPI(1),ZPRP(1,1))
 DATA ASYM /1HH,2HHE,2HLI,2HBE,1HB,1HC,1HN,1HO,1HF,2HNE,2HNA,2HMG,
 2HAL,2HSI,1HP,1HS,2HCL,2HAR,1HK,2HCA,2HSC,2HTI,1HV,2HCR,2HMN,
 2HFE,2HCO,2HNI,2HCU,2HZN,2HGA,2HGE,2HAS,2HSE,2HBR,2HCR,2HRB,
 2HFR,1HY,2HZR,2HNB,2HMO,2HTC,2HRU,2HRH,2HPD,2HAG,2HCD,2HIN,
 2HSN,2HSR,2HTE,1HI,2HXE,2HCS,2HBA,2HLA,2HCE,2HPR,2HND,2HPM,
 2HSM,2HEU,2HGD,2HIB,2HDY,2HHC,2HER,2HTM,2HYB,2HLU,2HMF,2HTA,
 1HW,2HRE,2HOS,2HIR,2HPT,2HAU,2HHG/
 DATA IUPDI /1/
 DATA PLUS /1H+/, ONE /1./
 DATA SHAPD /STRA,IGHT,.,LIN,.,E,.,CIRC,.,LE B,.,OTTO,.,M,.
 DATA CIRC,.,LE T,.,OP,.,/
 1 DATA SPNAME /SP1,.,SP2,.,SP3,.,SP4,.,SP5,.,SP6,.,SP7,.,SP8,.,
 .SP9,.,SP10,.,SP11,.,SP12,.,SP13,.,SP14,.,SP15,.,SP16,.,
 .SP17,.,SP18,.,SP19,.,SP20,.,SP21,.,SP22,.,SP23,.,SP24,.,
 .SP25,.,SP26,.,SP27,.,SP28,.,SP29,.,SP30./
 DATA SYI,SY2 /3*1H(,3*1H)/
 DATA WORD /6HTWO-DI,6HMENSIO,3HNAL,6H AXIS,6HYMMETR,2HIC/
 DATA WORD2 /3HGAP,4HDIAM/
 NAMELIST /INPUT/
 *****GROUP 1 - GENERAL CONTROL VARIABLES
 1 ISW1A,ISW2A,ISW3A,ISW4A,ISW6A,ISW1B,ISW3B,TWALL,NOTRAN,
 1 TSTOPI,CXMAXI,READG,READXS,
 *****GROUP 2 - OUTPUT CONTROLS
 2 ISW6H,ISW7B,TPRNTI,DATAP,NRECO,IRUN,
 *****GROUP 3 - RESERVOIR CONDITIONS
 3 ISW2B,PRESA,FLOW,CTAPI,HSTAG,MFITER,
 *****GROUP 4 - GEOMETRY
 4 NOZZLE,ICHAN,JDIM,NPRFLS,NPROFL,NBL,DIAM,NSECTS,ISHAPE,
 4 PARAMI,ATPI,XZEROI,
 *****GROUP 5 - GAS MODEL
 5 IGAS,AAMS,NCJCS,QPJ,ISCI,ISSI,ISRI,ICI,IE,IS,IR,ISATOM,
 5 ISMOL,CTMXXI,BZEROI,INEOVI,

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C *****GROUP 6 - TEST MODEL
C XMODP1,NMODPT,TSDIAM,AXIMOD,KDIM,r STAG,CATFAC,LEWIS,TMODEL,
C TFLATE,
C *****GROUP 7 - WEDGE MDELS
C ANGLE,ANGLE,NRADLE,RADLE,WX1,DWX,NWX,WXI,TWEDGE,WK,ISW9B,
C *****GROUP 8 - CONTROLS FOR THE FLOW SOLUTIONS
C WSAVE,DELTI,DELTXI,CCHI,NQSI,TTEST,GTEST,HTEST,TETEST,
C QTEST,GAMIN,DCHLL,DCHRAT,
C *****GROUP 9 - ELECTRONIC NONEQUILIBRIUM
C INT,KTF,KTR,ITR,EPAR,EPAR,TLIST,POM,
C *****GROUP 10 - CONTROLCLS FOR DIAGNOSTIC DUMPS
C ISWSA,ISW4B,ISW5B,ISW8B
C $ NAMELIST /EINPUT/
C EEP1,EEP2,EEP3,EEP4,EEPS,EEP5,EEP6,EEP7,EEP8,
C EEP9,EEP10,NEELS,IEFP,SP1,SP2,SP3,SP4,SP5,SP6,SP7,SP8,SP9,
C SP10,SP11,SP12,SP13,SP14,SP15,SP16,SP17,SP18,SP19,SP20,SP21,
C SP22,SP23,SP24,SP25,SP26,SP27,SP28,SP29,SP30,RP1,RP2,RP3,RP4,
C RP5,RP6,RP7,RP8,RP9,RP10,RP11,RP12,RP13,RP14,RP15,RP16,RP17,RP18,
C RP19,RP20,RP21,RP22,RP23,RP24,RP25,RP26,RP27,RP28,RP29,RP30,
C RP31,RP32,RP33,RP34,RP35,RP36,RP37,RP38,RP39,RP40,RP41,RP42,
C RP43,RP44,RP45,RP46,RP47,RP48,RP49,RP50,RP51,RP52,RP53,RP54,
C RP55,RP56,RP57,RP58,RP59,RP60,RP61,RP62,RP63,RP64,RP65,RP66,
C RP67,RP68,RP69,RP70,RP71,RP72,RP73,RP74,RP75,RP76,RP77,RP78,
C RP79,RP80,RP81,RP82,RP83,RP84,RP85,RP86,RP87,RP88,RP89,RP90,
C RP91,RP92
C $ NAMELIST /KKG,NNO,NKQ,ISEO,V1,V2,V3,V4,V5,V6,V7,V8,V9,V10,
C V11,V12,V13,V14,V15,V16,V17,V18,V19,V20,V21,V22,V23,V24,V25,V26,
C V27,V28,V29,V30,V31,V32,V33,V34,V35,V36,V37,V38,V39,V40,V41,V42,
C V43,V44,V45,V46,V47,V48,V49,V50,V51,V52,V53,V54,V55,V56,V57,V58,
C V59,V60,V61,V62,V63,V64,V65,V66,V67,V68,V69,V70,V71,V72,V73,V74,
C V75,V76,V77,V78,V79,V80,V81,V82,V83,V84,V85,V86,V87,V88,V89,V90,
C V91,V92,V93,V94,V95,V96,V97,V98,V99,V100,V101,V102,V103,V104,
C V105,V106,V107,V108,V109,V110,V111,V112,V113,V114,V115,V116,
C V117,V118,V119,V120,V121,V122,V123,V124,
C V125,V126,V127,V128,V129,V130,V131,V132,V133,V134,V135,V136,
C V137,V138,V139,V140,
C V141,V142,V143,V144,V145,V146,V147,V148,V149,V150,V151,V152,
C V153,V154,V155,V156,
C V157,V158,V159,V160,V161,V162,V163,V164,V165,V166,V167,
C V168,V169,V170,V171,V172,
C V173,V174,V175,V176,V177,V178,V179,V180,V181,V182,V183,
C V184,V185,V186,V187,
C V188,V189,V190,V191,V192,V193,V194,V195,V196,V197,V198,
C V199,V200,V201,V202,V203,V204,V205,V206,V207,V208,V209,
C V210,V211,V212,V213,V214,V215,V216,V217,V218,V219,V220,
C V221,V222,V223,V224,V225,V226,V227,V228,V229,V230,
C V231,V232,V233,V234,V235,V236,V237,V238,V239,V240,
C V241,V242,V243,V244,V245,V246,V247,V248,V249,V250,
C V251,V252,V253,V254,V255,V256,V257,V258,V259,V260,
C V261,V262,V263,V264,V265,V266,V267,V268,V269,V270,
C V271,V272,V273,V274,V275,V276,V277,V278,V279,V280,
C V281,V282,V283,V284,V285,V286,V287,V288,V289,V290,
C V291,V292,V293,V294,V295,V296,V297,V298,V299,V300,
C V301,V302,V303,V304,V305,V306,V307,V308,V309,V310,
C V311,V312,V313,V314,V315,V316,V317,V318,V319,V320,
C V321,V322,V323,V324,V325,V326,V327,V328,V329,V330,
C V331,V332,V333,V334,V335,V336,V337,V338,V339,V340,
C V341,V342,V343,V344,V345,V346,V347,V348,V349,V350,
C V351,V352,V353,V354,V355,V356,V357,V358,V359,V360,
C V361,V362,V363,V364,V365,V366,V367,V368,V369,
C V370,V371,V372,V373,V374,V375,V376,V377,V378,V379,
C V380,V381,V382,V383,V384,V385,
C V386,V387,V388,V389,V390,V391,V392,V393,V394,V395,
C V396,V397,V398,V399,V400,
C $ READ (5,96C) (ACOM(I),I=1,20)
C FIRST 4 CHARACTERS ON THIS CARD ARE USED AS A FACILITY NAME IF
C NOZZLE=0 (NONSTANDARD NOZZLE)
C READ (5,INPUT)
C WRITE (6,970)
C IF (.NOT.READG) GO TO 20
C READ (5,EINPUT)
C WRITE (6,EINPUT)
C DJ 10 I=1,20
C I: (SPRP(I,I),NE,0.) GO TO 10

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10 SPRP(1,I)=SPNAME(I)
20 CONTINUE
30 IF (.NOT.READ) GO TO 30
   READ (5,INPUT)
   WRITE (6,INPUT)
   CONTINUE
40 ICASE=ICASE+1
   WRITE (6,98C) IRUN,ICASE,(ACOM(I),I=1,20)
   LETIG=AAKS.AND.(IGAS.EQ.1.OR.IGAS.EQ.2)
   DO 40 I=1,4
   DUM(I)=C.
   IF (ICHAN.GT.0) NPRFLS=2
   IF (NPRFLS.EQ.1.OR.NPRFLS.EQ.2) GO TO 50
   WRITE (6,990) NPRFLS
50 STOP
   DO 60 I=1,NPRFLS
   IF (ICHAN.GT.0) NPROFL(I)=CP(I,ICHAN)
   IPRFL=NPROFL(I)
   IF (IPRFL.GT.0) DIAM(I)=ZPRP(1,IPRFL)/1.27
60 CONTINUE
   AXISYM=AXIMOD
   IF (NPRFLS.EQ.2) AXISYM=.FALSE.
   IF (ISW2B) 70,8C,170
   HS=HSTAG/1.8
   WRITE (6,1000) HSTAG,FLOW
70 GO TO 90
   WRITE (6,1010) PRESAI,FLOW
80 IF (NPRFLS.EQ.2) GO TO 110
90 IF (JDIM.NE.0) GO TO 100
   SMASS=70.3069*FLOW/DIAM(I)
   GO TO 120
100 SMASS=89.5173*FLOW/(DIAM(I))**2
   GO TO 120
110 SMASS=70.3069*FLOW/DIAM(I)/DIAM(2)
120 IF (.NOT.SETIG) GO TO 180
   IF (ISW2B.EQ.0) GO TO 130
   HWINOV=HSTAG
   GO TO 14
130 HWINOV=(-S.7*PRESAI/SMASS)**2.52
140 IF (HWINOV-8000.) 150,150,160
150 IGAS=2
   GO TO 180
160 IGAS=1
   GO TO 180
170 WRITE (6,1020) PRESAI,CTAPI
   IF (.NOT.SETIG) GO TO 180
   IF (CTAPI-6000.) 150,150,160
180 IUPD=IUPDI
   NQS=NQSI
   IF (FSTAG.NE.0.) FSTAG=SIGN(ONE,FSTAG)
   RSA=1.
   IF (NPRFLS.EQ.2) GO TO 190
   NBL=1
   GO TO 200
  
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19C IF ( ICHAN.GT.0) NBL=CP(4, ICHAN)
200 IF ( ISWB.NF.0) ISWZB=1
    LIMOUT=ISW7B*ISW3B
    INEQV=INEQVI
    CTAP=CTAPI
    PRESA=PRESAI
    CTMX=CTMXXI
    CXMAX=CXMAXI*2.5
    MBL=1
    IF (NPRFLS.EQ.1) GO TO 210
    IF (NRL.EQ.1) MBL=2
    NTS=0
    DD 230 I=1,20
    IF (TSDIAM(I).LE.DIAM(MBL)) TSDIAM(I)=1.E30
    IF (TSDIAM(I).GT.9.E19) GO TO 230
    NTS=NTS+1
    IF (NPRFLS.EQ.2) GO TO 220
    TSAR(NTS)=(TSDIAM(I)/DIAM(1))*((JDI'+1)
    GO TO 230
    TSAR(NTS)=1.27*TSDIAM(I)
    CONTINUE
    IF (ANGLE.EQ.C.OR.NRADLE.EQ.0) GO TO 240
    IF (NXX.EQ.0.AND.WXI(1).GT.1.E19) GO TO 240
    IF (NPRFLS.EQ.2) GO TO 240
    WEDGFM=.TRUE.
    GO TO 250
    WEDGFM=.FALSE.
    IF (NOZZLE.GT.0) GO TO 260
    XZERO=2.54*XZEROI
    GO TO 270
    XZERO=ZPRP(2,NOZZLE)
    IN THE CASE OF A CHANNEL, THE CODE USES THE XZERO FOR THE FIRST
    PROFILE
    SL=1.
    BZERO=BZEROI
    TSTOP=TSTOPI
    TPRINT=TPRNTI
    DELTAX=DELTXI
    DELTI=DELTI1
    DD 320 J=1,NPRFLS
    IPRFL=NPROFL(J)
    IF (IPRFL.EQ.0) GO TO 300
    RTHCM(J)=ZPRP(1,IPRFL)
    NSECTU(J)=ZPRP(3,IPRFL)
    NSECTD(J)=ZPRP(4,IPRFL)
    NSECT(J)=NSECTU(J)+NSECTD(J)
    IF (J.EQ.1) FACNAM=ZPRP(64,IPRFL)
    NSECTJ=NSECT(J)
    DD 290 I=1,NSECTJ
    ISHAPE(I,J)=ZPRP(I+4,IPRFL)
    IF (I.LT.NSECTJ) ATPI(I,J)=ZPRP(I+16,IPRFL)
    DD 280 L=1,3
    LL=3*I+L+24
    PARAM(L,I,J)=ZPRP(LL,IPRFL)
    CONTINUE
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300 GO TO 320
    NSECTU(J)=NSECTS(1,J)
    NSECTD(J)=NSECTS(2,J)
    NSECT(J)=NSECTU(J)+NSECTD(J)
    IF (J.EQ.1) FACNAM=ACOM(1)
    RTHCM(J)=1,27*DIAM(J)
    NSECTJ=NSECT(J)
    DO 310 I=1,NSECTJ
    DO 310 L=1,3
    PARAM(L,I,J)=PARAMI(L,I,J)
CONTINUE
    IF (ICHAN.EQ.0) GO TO 330
    CHANAM=CP(3,ICHAN)
    FACNAM=CP(5,ICHAN)
    GO TO 340
    CHANAM=ACOM(2)
CONTINUE
310 IF (NPRFLS.EQ.1) GO TO 360
    IF (ICHAN.NE.0) GO TO 350
    WRITE (6,1030) CHANAM,FACNAM,DIAM(1),DIAM(2)
    GO TO 390
350 WRITE (6,1040) CHANAM,ICHAN,FACNAM,DIAM(1),DIAM(2)
    GO TO 390
360 JDPI=JDIM+1
    IF (NOZZLE.GT.0) GO TO 370
    WRITE (6,1050) (WORD(I,JDPI),I=1,3),DIAM(1),WORD2(JDPI),FACNAM
    GO TO 380
370 WRITE (6,1060) (WORD(I,JDPI),I=1,3),NOZZLE,DIAM(1),WORD2(JDPI),FACNAM
    1NAV
380 IF (NOZZLE.EQ.NOZZO.AND.NOZZLE.NE.0) GO TO 440
    GO TO 400
390 IF (ICHAN.EQ.ICHANO.AND.ICHAN.NE.0) GO TO 440
    DO 430 J=1,NPRFLS
    IF (NPRFLS.EQ.1) GO TO 410
    WRITE (6,1070) J
    WRITE (6,1080) RTHCM(J),XZERO,NSECT(J),NSECTU(J)
    WRITE (6,1090)
    NSM1=NSECT(J)-1
    DO 420 K=1,NSM1
    ISH=ISHAPE(K,J)
    WRITE (6,1100) K,ISH,(SHAPD(I,ISH),I=1,4),ATPI(K,J),(PARAM(L,K,J),
    1L=1,3)
    NSECTJ=NSECT(J)
    K=NSECTJ
    ISH=ISHAPE(NSECTJ,J)
    WRITE (6,1110) NSECTJ,ISH,(SHAPD(I,ISH),I=1,4),(PARAM(L,K,J),L=1,3)
    1)
430 CONTINUE
440 CONTINUE
    IF (NEELS.EQ.0) GO TO 460
    DO 450 I=1,NEELS
    IZ=EEPRP(1,I)+0.1
    II=IEEP(I)
    EPRP(1,II)=ASYM(IZ)
    EPRP(2,II)=EEPRP(2,I)
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460 IG=IGAS
    IF (IGAS.NE.0) GO TO 470
    WRITE (6,1120)
    ISC=ISCI
    ISS=ISSI
    ISR=ISRI
    IC=ICI
    GO TO 520
470 IGASE=ABS(IGAS)
    WRITE (6,1130) IGAS,GPRP(1,IGAS)
    SET UP DESCRIPT JN OF GAS IN TERMS OF ELEMENTS, SPECIES, AND
    C REACTIONS
    ISC=GPRP(2,IGAS)
    ISS=GPRP(3,IGAS)
    ISR=GPRP(4,IGAS)
    IC=GPRP(5,IGAS)
    DO 490 I=1,ISC
    IE(I)=GPRP(I+5,IGAS)
    NCS=GPRP(120,IGAS)
    DO 490 I=1,NCS
    IF (IGSLT.0) GO TO 490
    OPJ(I)=GPRP(I+15,IGAS)
    JCS(I)=GPRP(I+109,IGAS)
    DO 500 I=1,ISS
    IS(I)=GPRP(I+25,IGAS)
    DO 510 I=1,ISR
    IR(I)=GPRP(I+45,IGAS)
    ISATOM=GPRP(121,IGAS)
    ISMOL=GPRP(122,IGAS)
    INT=GPRP(123,IGAS)
    LEWIS=GPRP(124,IGAS)
    IF (INT.EQ.0) GO TO 530
    NT=2
    GO TO 540
530 NT=1
540 CONTINUE
    LOOK UP ELEMENTS
    DO 550 I=1,ISC
    II=IE(I)
    ELEMENT(II)=EPRP(1,II)
    CMW(II)=EPRP(2,II)
    LOOK UP SPECIES
    IAMBIP=1
    NFIT=0
    DO 680 I=1,ISS
    II=IS(I)
    DO 560 K=1,NCS
    IF (II.EQ.JCS(K)) GO TO 570
    CONTINUE
    GO TO 580
560 IJCS(K)=I
570 CONTINUE
    JATOM=I
580 IF (II.EQ.ISMOL) JMOL=I
    HP(I)=SPRP(1,II)
  
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NELS=SPRP(2,II)+0.1
DO 630 K=1,ISC
DO 590 J=1,NELS
IES=SPRP(J+2,II)+0.1
IF (IE(K).EQ.IES) GO TO 600
CONTINUE
LPIJ(I,K)=0
GO TO 630
AL=SPRP(J+5,II)
LP=AL
DAL=AL-LP
IF (ABS(DAL).LE.0.5) GO TO 620
IF (DAL.GT.0.) GO TO 610
LP=LP-1
GO TO 620
LP=LP+1
LPIJ(I,K)=LP
IF (II.EQ.ISATOM.AND.IES.EQ.1) IAMBIP=2
CONTINUE
TFA(I)=SPRP(9,II)
TFB(I)=SPRP(10,II)
TFC(I)=SPRP(11,II)
TFD(I)=SPRP(12,II)
TFE(I)=SPRP(13,II)
TFK(I)=SPRP(14,II)
SHJAP(I)=SPRP(15,II)
ETAJ(I)=SPRP(16,II)
SBJ(I)=SPRP(17,II)
THEVP(I)=SPRP(18,II)
IF (ETAJ(I).LT.2.9) GO TO 650
DO 640 K=1,3
THEVE(K+1,II)=SPRP(K+40,II)
IGM(I)=SPRP(19,II)+0.1
IGJ(I)=SPRP(20,II)+0.1
IF (IGJ(I).NE.0) NFIT=1
IF (ETAJ(I).NE.C.OR.IGJ(I).NE.0) GO TO 660
WRITE (6,II90) I,II
STOP
IGMV=IGM(I)
DO 670 L=1,IGMV
GELJ(L,II)=SPRP(L+20,II)
ELJ(L,II)=SPRP(L+30,II)
CONTINUE
DO 700 I=1,ISC
CAPQ(I)=0.
DO 700 J=1,NCS
JJ=JCS(J)
NELS=SPRP(2,JJ)+0.1
DO 690 K=1,NELS
KK=SPRP(K+2,JJ)+0.1
IF (KK.NE.IE(I)) GO TO 690
CAPQ(I)=CAPQ(I)+OPJ(J)*SPRP(K+5,JJ)
CONTINUE
CONTINUE
LOOK UP REACTIONS

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DO 780 I=1,ISR
II=IR(I)
CAI(I)=RPRP(1,II)
ETA(I)=RPRP(2,II)
CEACT(I)=RPRP(3,II)
QQ(I)=RPRP(4,II)
NPSS=RPRP(5,II)+0.1
DO 770 J=1,ISS
DO 710 K=1,NR3S
JRS=RPRP(K+6,II)+0.1
IF (JRS.NE.IS(J)) GO TO 710
XNUIJ(I,J)=RPRP(K+12,II)
GO TO 720
CONTINUE
XNUIJ(I,J)=0.
CONTINUE
DO 730 K=1,NPSS
JPS=RPRP(K+9,II)+0.1
IF (JPS.NE.IS(J)) GO TO 730
XNUIJP(I,J)=RPRP(K+15,II)
GO TO 740
CONTINUE
XNUIJP(I,J)=0.
CONTINUE
NK=RPRP(19,II)+0.1
DO 750 K=1,NK
ISKUR=RPRP(K+19,II)+0.1
IF (ISKUR.EQ.IS(J)) GO TO 760
CONTINUE
KUR(I,J)=0
GO TO 770
KUR(I,J)=1
CONTINUE
IF (IGAS.EQ.0.OR.NT.EQ.1) GO TO 780
KTR(I)=TNEP(I,INT)
ITR(I)=TNEP(I+25,INT)
EPAR(1,I)=TNEP(I+50,INT)
EPAR(2,I)=TNEP(I+75,INT)
CONTINUE
IF (NT.EQ.1) GO TO 800
IF (IGAS.EQ.0) GO TO 800
DO 790 I=1,30
TLST(I)=TNEP(I+125,INT)
POM(I)=TNEP(I+155,INT)
BPAR=TNEP(186,INT)
DO 820 I=2,ISS
IF (LPIJ(I,1).EQ.0) GO TO 810
ICH(I)=2
GO TO 820
ICH(I)=1
CONTINUE
DO 850 I=1,ISR
IF (KTF(I).NE.4) GO TO 840
  
```

710
 720
 730
 740
 750
 760
 770
 780
 790
 800
 810
 820

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```
830 DD 830 J=2,ISS
840 IF (XNUIJP(I,J).EQ.0.) GO TO 830
850 IPA(I)=J
860 GO TO 850
CONTINUE
870 IPA(I)=0
CONTINUE
IZERO=0
SUPGO=IGAS.NE.0.AND.IGAS.EQ.IGASD
IF (SUPGO) GO TO 870
WRITE (6,1140)
WRITE (6,1150)
CGMW=C
DD 890 I=1,NCS
JJ=JCS(I)
NELS=SPRP(2,JJ)
CSMW=0
DD 880 K=1,NELS
KK=SPRP(K+2,JJ)
ELNAME(K)=EPRP(1,KK)
NATOM(K)=SPRP(K+5,JJ)
CSM+=CSMW+NATOM(K)*EPRP(2,KK)
IF (KK.NE.1) GO TO 880
ELNAME(K)=PLUS
NATOM(K)=-NATOM(K)
CONTINUE
880 CGMW=CGMW+CSMW*OPJ(I)
IF (SUPGO) GO TO 890
WRITE (6,1160) I,SPRP(1,JJ),JJ,OPJ(I),CSMW,(SY1(K),ELNAME(K),SY2(K)
I),NATOM(K),K=1,NELS)
CONTINUE
890 IF (.NOT.SUPGO) WRITE (6,1170) CGMW
IGASO=IGAS
NOZZO=NOZZLE
ICHANO=ICHAN
IGAS=IGS
IF (XMODP1) 900,900,910
900 XMODP1=1.E20
WRITE (6,1180)
910 XMPI=XMODP1*2.54
IF (CXMAXI.GT.XMODP1.AND.NMODPT.GT.1) GO TO 920
FACMP=1.E10
920 GO TO 930
ENMP=NMODPT-1
EXMP=1./FNMP
FACMP=(CXMAXI/XMODP1)*EXMP
930 CONTINUE
IF (NRECO.LE.0.OR.ICASE.GT.1) GO TO 950
940 DD 940 I=1,NRECO
950 READ (ITPUT)
CONTINUE
RETURN
C
C
C
```

960 FORMAT (20A4) REA 660
 970 FORMAT (1H1) REA 661
 980 FORMAT (1H1,55X,20HNATA III CODE OUTPUT/8HORUN NO.,17,7X,4HCASE,14 REA 662
 1,12H IN THIS JOB,9X,20A4//) REA 663
 990 FORMAT (30H0 INVALID INPUT DATA.,. NPRFLS=,I10) REA 664
 1000 FORMAT (21H STAGNATION ENTHALPY=,F8,0,25H BTU/LB, TOTAL MASS FLOW= REA 665
 1, F9,5,7H LB/SEC) REA 666
 1010 FORMAT (20H RESERVOIR PRESSURE=,F9,4,23H ATM, TOTAL MASS FLOW =,F9 REA 667
 1,5,7H LB/SEC) REA 668
 1020 FORMAT (20H RESERVOIR PRESSURE=,F9,4,28H ATM, RESERVOIR TEMPERATUR REA 669
 1E=,F9,2,6H DEG K) REA 670
 1030 FORMAT (/13H NONSTANDARD ,A4,13H CHANNEL FOR ,A4,21I, ,F7,4,3H BY, REA 671
 1F7,4,12H INCH THROAT) REA 672
 1040 FORMAT (/1X,A4,30H CHANNEL (STANDARD CHANNEL NO.,12,5H) FOR,1X,A4 REA 673
 1,2H, ,F7,4,3H BY, F7,4,12H INCH THROAT) REA 674
 1050 FORMAT (/13H NONSTANDARD ,2A6,A3,8H NOZZLE,,F7,3,13H INCH THROAT REA 675
 1,A4,5H FOR ,A4) REA 676
 1060 FCFORMAT (/10H STANDARD ,2A6,A3,11H NOZZLE NO.,13,1H,,F7,3,13H INCH REA 677
 1 THROAT ,A4,5H FOR ,A4) REA 678
 1070 FORMAT (/60X,11H PROFILE NO.,12) REA 679
 1080 FORMAT (15H THROAT RADIUS=,F7,3,3H CM,5X,9H INLET AT,F8,3,3H CM,5X, REA 680
 112,17H SECTIONS IN FIT,,12,19H UPSTREAM OF THROAT) REA 681
 1090 FORMAT (1H0,3X,1HJ,6X,9H SHAPE(J),15X,5H SHAPE,13X,7H ATPI(J),10X,10 REA 682
 1HPARAM(1,J),10X,10H PARAM(2,J),10X,10H PARAM(3,J)) REA 683
 1100 FORMAT (15,115,4X,4A4,1P4E20,4) REA 684
 1110 FORMAT (15,115,4X,4A4,20X,1P3E20,4) REA 685
 1120 FCFORMAT (/24H NONSTANDARD GAS MIXTURE) REA 686
 1130 FCFORMAT (/18H STANDARD GAS NO, ,12,2H (,A6,1H)) REA 687
 1140 FCFORMAT (/56X,12H COLD SPECIES) REA 688
 1150 FCFORMAT (1H0,16X,3HNO,,4X,4HNAME,7X,5H INDEX,10X,10H MOLE FRAC.,12X,8 REA 689
 1HMOL, WT.,14X,16H CHEMICAL FORMULA) REA 690
 1160 FCFORMAT (10X,110,5X,A5,110,F20,5,F20,4,15X,3(A1,A2,A1,11,1X)) REA 691
 1170 FCFORMAT (35H0 MEAN MOLECULAR WEIGHT OF COLD GAS=,F9,4//) REA 692
 1180 FCFORMAT (61H0 NEGATIVE OR ZERO VALUES OF XMODPI NOT ALLOWED, DATA IG REA 693
 1NORED.) REA 694
 1190 FCFORMAT (49H0 NO THERMAL PROPERTY DATA DEFINED FOR SPECIES NO.,13,30 REA 695
 1H IN THE CURRENT GAS MODEL (NO.,13,31H IN THE MASTER LIST OF SPECIE REA 696
 2ES)) REA 697
 END REA 698-

SUBROUTINE RESET
SIMULATION OF UNIVAC 1108 TIMING ROUTINES RESET AND TIME

C
IN=1
GO TO 10
ENTRY TIME(I)
IN=2
CALL ACUCPU (ICPU)
GO TO (20,30), IN
IZERO=ICPU
GO TO 40
I=IZERC-ICPU
RETURN
END
10
20
30
40

1 RST
2 RST
3 RST
4 RST
5 RST
6 RST
7 RST
8 RST
9 RST
10 RST
11 RST
12 RST
13- RST

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RES 56
RES 57
RES 58
RES 59
RES 60
RES 61
RES 62
RES 63
RES 64
RES 65
RES 66
RES 67
RES 68
RES 69
RES 70
RES 71
RES 72
RES 73
RES 74
RES 75
RES 76
RES 77
RES 78
RES 79
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RES 81
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RES 98
RES 99
RES 100
RES 101
RES 102
RES 103
RES 104
RES 105
RES 106
RES 107
RES 108
RES 109
RES 110

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N=N+1
IF (N.GT.20) GO TO 160
DO 20 I=1,2
J=3-I
CTSAVE(J+1)=CTSAVE(J)
F(J+1)=F(J)
CALL INIT
CALL INTA
IF (ERR) RETURN
CALL NRMVX
IF (ERR) RETURN
CTSAVE(1)=CTAP
F(1)=CMA*SM*SM
IF (ISW5A.NE.0) WRITE (6,RSTOMP)
IF (N-2) 30,40,50
ITERATE TO GET SECOND POINT
CTAPEF(1)/CONST
GO TO 10
LINEAR INTERPOLATION TO GET THIRD POINT
DCT=CTSAVE(2)-CTSAVE(1)
D1=(F(1)*CTSAVE(2)-F(2)*CTSAVE(1))/DCT
D2=(F(2)-F(1))/DCT
CTAPE=-D1/(D2-CONST)
DCTAPO=ABS(CTAP-CTSAVE(1))
IF (DCTAPO/ABS(CTAP).LE.TERR) GO TO 190
GO TO 10
QUADRATIC INTERPOLATION
DO 60 I=1,3
AM(I,1)=1.00000000
AM(I,2)=CTSAVE(I)
AM(I,3)=CTSAVE(I)*CTSAVE(I)
BM(I)=F(I)
CALL SIMQ (AM,BM,3,KS)
IF (KS.NE.0) GO TO 40
DMC=BM(2)-CONST
IF (BM(3).NE.0.) GO TO 70
CTAPE=-BM(1)/DMC
GO TO 80
CTAPE=(-DMC+SIGN(SORT(DMC*DMC-4.*BM(1)*BM(3)),DMC))*0.5/BM(3)
DCTAP=ABS(CTAP-CTSAVE(1))
IF (DCTAP/ABS(CTAP).LE.TERR) GO TO 190
IF (DCTAP.GE.DCTAPO) GO TO 40
DCTAPO=DCTAP
GO TO 10
ENTHALPY-MASS FLOW OPTION
PRESA=4.8E-3*SMASS*SORT(HS)
HEMR=1.E-5*HS
SMERR=1.E-5*SMASS
CTAPE=10000.
N=N+1
IF (N.GT.20) GO TO 160
PLAST=CTAP
PLAST=PRESA
N2=0
N2=N2+1

```

10
20
C 30
C 40
C 50
60
70
80
C 90
100
110

SUBROUTINE RNKT
 LOGICAL ERR, FAILED, TE
 REAL S1, S2, S3, T1, T2, T3, Z
 REAL ACOM(30), ELEMENT(10), HP(20)
 DOUBLE PRECISION AA, AAA, CAPX, GJ, CDIJ
 COMMON AA(22, 24), CDIJ(20, 10), CAPX(20)
 COMMON A, AFNTS, AFNX, AMACH, AR, ARBA, ARBB, BZERO,
 C, CARB, CH, CHA, CLNT, CM, CMA, CRA, CRP,
 1 CRRB, CRS, CSTA, CT, CTAP, CTB, CTC, CTMAX, CTMX,
 2 CTP, CTPL, CTT, CX, CX3, CXMAX, DATEST, DBTEST, DELTI,
 4 DELT2, DELTAX, DLOGA, DLOGR, PRESTH, PRHO, RHAP, RHO, RHOB,
 5 SPCTEST, PRES, PRESA, RHOP, RHPL, RHTH, ROBARA, ROBARP, SCPC, SDT,
 6 RHOBAR, RHOC, RHOD, RHP, RHP, RHTH, ROBARA, ROBARP, SCPC, SDT,
 7 SEN, SHPG, SC, SL, SL64, SM, SU, SUMG,
 8 TEST, TESTB, TPRINT, TSTOP, UP, ZP, ZPA
 COMMON BE(64), BLBK(31), CAI(64), CAPXTH(20),
 1 CCRPJ(20), CEACT(64), CGI(20), CGMU(20), CHI(64), CHI(20),
 2 CLN1, CLN2, CLNPI(64), CMW(20), ETAI(64), ETAJ(20), GJA(20),
 3 GJB(20), PERTGJ(20), PGJ(20), PI(64), PACHI(64), QM(20),
 4 QU(64), SAJ(20), SDCHI(64), SENT(20), SHJ(20), SHJA(20),
 5 SKIL(20), SS(20), TB(20), TFA(20), TF(20), TF(20),
 6 TFD(20), TFE(20), TFK(20), THEV(20), XMJAT(20), XNU(64),
 COMMON BETA(64, 20), ELJ(10, 20), GELJ(10, 20),
 1 XNUIJ(64, 20), XNUIJP(64, 20)
 COMMON IC, INEQ, INEQV, IP, IRUN, ISC, ISCP1,
 1 ISMC, ISMCNR, ISR, ISS, ISSNR, ISSP1, ISSP2, ISSP3, ISSP4,
 2 ISWA, ISW1G, ISW2A, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
 3 ISW5B, ISW6A, ISW6B, IUPD, IZER, JJK, LC, MI, NFIT,
 4 NIT, NNN, NNS, NQS, NOT, NTEST
 COMMON IGM(20), ITR(5), KUR(64, 20), LPIJ(20, 10)
 COMMON ACOM, ELEMENT, HP
 DIMENSION AAA(22, 24), BTA(64, 20), CAPO(31), CCI(20), DGJ(20), GJ(20), SBRNK
 1 J(20), SDGJ(20), SHJAP(20), THEVP(20)
 EQUIVALENCE (AAA(1, 1), AAA(1, 1)), (BETA(1, 1), BTA(1, 1)),
 (BLBK(1), CAPO(1)), (CAPXTH(1), CCI(1)), (GJA(1), DGJ(1)),
 2 (CAPX(1), GJ(1)), (SAJ(1), SBJ(1)), (SS(1), SDGJ(1)),
 3 (SHJA(1), SHJAP(1)), (THEV(1), THEVP(1))
 COMMON /ERROR/ ERR
 COMMON /TNERK/ SDTE, CTB, DCHA, CHB, SDCHA, DQMAX, IFAIL
 COMMON /CERROR/ FAILED
 COMMON /TNONEQ/ CTE, DTE, BPAR, EPAR(2, 25), NT, ITR(25), KTF(25),
 1 KTR(25), ICH(20), IPA(25)
 COMMON /TNCE/ SUMGH, SCPGH, CDPR, QDPE
 DIMENSION J1(23), GJ2(23), GJ3(23), GJ4(23), F1(23), F2(23), F3(23),
 1 F4(23), X1(23), X2(23), X3(23), X4(23), P(23), SDQ(23), FK(23, 4),
 2 GJK(23, 4), Y(4)
 EQUIVALENCE (Y1, Y(1)), (Y2, Y(2)), (Y3, Y(3)), (Y4, Y(4))
 EQUIVALENCE (SDQ(1), AA(1, 1))
 EQUIVALENCE (FK(1, 1), F1(1)), (FK(1, 2), F2(1)), (FK(1, 3), F3(1)),
 1 (FK(1, 4), F4(1)), (GJK(1, 1), GJ1(1)), (GJK(1, 2), GJ2(1)),
 2 (GJK(1, 3), GJ3(1)), (GJK(1, 4), GJ4(1))
 NAMELIST /RNKDMPI/ CTB, CXB, CT, CX, DT, HDELX, DLOGA, DLOGR, SU, AFNX,
 1 PRES, RHO, GJB, CGJ, CTE, DTE, DCHA
 NAMELIST /RNKDP2/ SDT, SDTE, SDCHA, SDGJ, F1, F2, F3, F4, P
 IFAIL=0

1 RNK
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 3 RNK
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 54 RNK
 55 RNK

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TE=NT.EQ.1
LIM=ISSPI
IF (.NOT. TE) LIM=LIM+2
QDPER=CDPE
DQMRK=2.*DQMAX
IF (ISWSB.EQ.0) GO TO 10
ISTMNT=1
WRITE (6,410) ISTMNT
DO 20 J=1,ISS
F1(J)=DGJ(J)
GJ1(J)=GJR(J)
GJ2(J)=GJ(J)
GJ2(J)=GJ(J)
F1(ISSPI)=DT
GJ1(ISSPI)=CTB
CT=CTB+DT*HDELX
GJ2(ISSPI)=CT
IF (TE) GO TO 30
F1(ISSP2)=DTE
GJ1(ISSP2)=CTEB
CTE=CTEB+DTE*HDELX
GJ2(ISSP2)=CTE
F1(ISSP3)=DCHA
GJ1(ISSP3)=CHR
CHA=CHB+DCHA*HDELX
GJ2(ISSP3)=CHA
DO 40 J=1,LIM
IFAIL=-J
IF (GJ2(J)) 270,40,40
CCNTINUF
IFAIL=0
CALL DERIVS
IF (ERR.OR.FAILED) RETURN
IF (ISWSB.EQ.0) GO TO 50
ISTMNT=2
WRITE (6,410) ISTMNT
WRITE (6,RNKDMP)
DO 60 J=1,ISS
F2(J)=DGJ(J)
GJ3(J)=GJ(J)
GJ3(J)=GJ(J)
F2(ISSPI)=DT
CT=CTB+DT*HDELX
GJ3(ISSPI)=CT
IF (TE) GO TO 70
F2(ISSP2)=DTF
CTE=CTEB+DTE*HDELX
GJ3(ISSP2)=CTE
F2(ISSP3)=DCHA
CHA=CHB+DCHA*HDELX
GJ3(ISSP3)=CHA
DO 80 J=1,LIM
IFAIL=-J

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RNK 56
RNK 57
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RNK 109
RNK 110

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80 IF (GJ3(J)) 270.80.80
CONTINUE
IF (TE) GO TO 90
IFAIL=ISSP4
IF (ABS(QDPE-QCPEB).LE.DQMRK.OR.DOMAX.EQ.0.) GO TO 90
GO TO 270
IFAIL=0
CALL DERIVS
IF (ERR.OR.FAILED) RETURN
IF (ISWSB.EQ.0) GO TJ 100
ISTMNT=3
WRITE (6.410) ISTMNT
WRITE (6.RNKDMP)
DO J=1,ISS
F3(J)=DGJ(J)
F3(ISSP1)=DT
IF (TE) GO TO 120
F3(ISSP2)=DTE
F3(ISSP3)=DCHA
DO I=1,LIM
IF (F1(J).EQ.0.) GO TO 130
X4(J)=F2(J)-F1(J)
IF (ABS(X4(J)/F1(J)).GE.0.001) GO TO 140
P(J)=0.0
GO TO 150
XX=ABS(F3(J)-F2(J))
IF (XX.EQ.0.) GO TO 130
XX=ALOG10(XX)
YY=ALOG10(DABS(X4,J))
IF ((XX-YY).GT.38.0) GO TO 130
P(J)=2.C*(F3(J)-F2(J))/X4(J)
CONTINUE
DO 210 J=1,LIM
IF (P(J)) 160,210,210
IF (P(J)+1.25) 200,200,170
Z=3.0
S1=C.C
S2=C.C
S3=C.C
T1=1.0
T2=0.5
T3=1.C/6.0
S1=S1+T1
S2=S2+T2
S3=S3+T3
Z=Z+1.0
T1=T1*P(J)/(Z-2.0)
T2=T2*P(J)/(Z-1.0)
T3=T3*P(J)/7
IF ((S3+T3)-S3) 180,190,180
X1(J)=S1
X2(J)=S2
X3(J)=S3
GO TO 210
X1(J)=(EXP(P(J))-1.C)/P(J)

```

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RNK 111
RNK 112
RNK 113
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RNK 162
RNK 163
RNK 164
RNK 165

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210 X2(J)=(X1(J)-1.0)/P(J)
X3(J)=(X2(J)-0.5)/P(J)
CONTINUE
DO 240 J=1,LIM
IF (P(J)) 230,220,220
G4(J)=GJ1(J)+F3(J)*DELTA
IFOR=1
GO TO 240
220 GJ4(J)=GJ1(J)+DELTA*(2.0*F3(J)*X2(J)+F1(J)*(X1(J)-2.0*X2(J))-F2(J)
1)*F(J)*X2(J))
IFOR=2
CONTINUE
DO 250 J=1,LIM
IFAIL=-J-30*IFOR
IF (GJ4(J)) 270,250,250
CONTINUE
IF (TE) GO TO 260
IFAIL=-ISSP4-30*IFOR
IF (ABS(QDPE-QDPEB).LE.DQMRK.OR.DQMAX.EQ.0.) GO TO 260
GO TO 270
260 IFAIL=0
GO TO 280
270 FAILED=.TRUE.
GO TO 400
280 DO 290 J=1,ISS
GJ(J)=GJ4(J)
CT=GJ4(ISSP1)
IF (TE) GO TO 300
CTE=GJ4(ISSP2)
CHA=GJ4(ISSP3)
CX=CX8+DELTA
CALL DEPIVS
IF (EAF.EQ.0) RETURN
IF (ISSP1.EQ.0) GO TO 310
ISTMT=1
WRITE (410) ISTMT
WRITE (6,RNKDMP)
DO 320 J=1,ISS
F4(J)=GJ(J)
F4(ISSP1)=DT
IF (TE) GO TO 330
F4(ISSP2)=DTE
F4(ISSP3)=DCHA
DO 370 J=1,LIM
IF (P(J)) 350,340,340
SDQ(J)=(DELTA/6.0)*(F1(J)+2.0*F2(J)+2.0*F3(J)+F4(J))
GO TO 370
DO 360 K=1,4
Y(K)=F(K,J)-(P(J)*GJK(J,K))/DELTA
SDQ(J)=DELTA*(F1(J)*X1(J)+(-3.0*Y1+2.0*Y2+2.0*Y3-Y4)*X2(J)+4.0*(Y
11-Y2-Y3+Y4)*X3(J))
CONTINUE
DO 380 J=1,ISS
SDGJ(J)=SDQ(J)
SDT=SDQ(ISSP1)

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RNK 166
RNK 167
RNK 168
RNK 169
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RNK 219
RNK 220

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IF (TE) GO TO 390
SDTE=SDQ(ISSP2)
SDCHA=SDQ(ISSP3)
IF (ISW58.EQ.0) GO TO 400
ISTMNT=5
WRITE (6,410) ISTMNT
WRITE (6,RNKDMP)
WRITE (6,RNKDP2)
RETURN

390

400

C

C

C

410

FORMAT (18H DUMP AT STATEMENT,12)
END

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RNK 222
RNK 223
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RNK 233
RNK 234-

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C      SUBROUTINE SHOCK (BMF,DEL,GAM,FL,IERR)
C      KEEP THIS IMPLICIT STATEMENT IN THE UN. AC VERSION
C      IMPLICIT REAL*8 (A-H,O-Z)
C      HARVEY BUSS ROUTINE - OBLIQUE SHOCK IN A PERFECT GAS
C
C      *****DEFINITIONS OF ARGUMENTS*****
C      BMF=FREE STREAM MACH NUMBER
C      DEL=WEDGE HALF ANGLE (DEGREES)
C      GAM=RATIO OF SPECIFIC HEATS
C      FL(1)=SHOCK ANGLE (RADIAN)
C      FL(2)=MACH NUMBER BEHIND SHOCK
C      FL(3)=DENSITY RATIO, RHO(BEHIND)/RHO(INFINITY)
C      FL(4)=STATIC PRESSURE RATIO, P(BEHIND)/P(INFINITY)
C      FL(5)=STATIC TEMPERATURE RATIO, T(BEHIND)/T(INFINITY)
C      FL(6)=TOTAL PRESSURE RATIO, PT(BEHIND)/PT(INFINITY)
C      FL(7)= TSTAG/TINF
C      IERR=1 NORMAL RETURN, IERR=2 ERROR RETURN, IERR=0 DETACHED SHOCK
C
C      DIMENSION FL(7)
C      KIT=0
C      DELR=DEL/57.295779
C      SD=SIN(DELR)
C      CD=COS(DELR)
C      BM2=BMF**2
C      BM4=BM2**2
C      C1=GAM+1.0
C      C2=GAM-1.0
C      B=-((BM2+2.0)/BM2-GAM*SD**2)
C      C=(2.*4*2+1.)/BM4+(C1**2/4.+C2/RM2)*SD**2
C      C3=2.*B
C      UG=0.
C      FU=CD+UG*(C+UG*(B+UG))
C      FUE=C+UG*(C3+3.*UG)
C      UN=UG-FU/FPU
C      IF (DABS(UG/UN-1.00)-1.0-8) 50.50.20
C      KIT=KIT+1
C      UG=UN
C      GO TO 10
C      WRITE (6,100)
C      IERR=2
C      RETURN
C      ALP=B+UN
C      BETA=UN*ALP+C
C      DIS=ALP**2-4.*BETA
C      IF (DIS.GE.0.) GO TO 60
C      WRITE (6,110)
C      IERR=0
C      RETURN
C      DIS=SQRT(DIS)
C      U1=(-ALP+DIS)/2.0
C      U2=(-ALP-DIS)/2.0
C      IF (U1-U2) 70.70.80
C      RT=U1

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SHK 55

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80      GO TO 90
C      RT=U2
90      80 ZT= SORT(RT)
C      CONTINUE
C      FL(1)=ATAN(ZT/SQRT(1.-ZT**2))
C      V1=BM2*RT
C      V2=C1**2*BM4*RT-4.0 *(V1-1.0 )*(GAM*V1+1.0 )
C      V3=(2.0 *GAM*V1-C2)*(C2*V1+2.0)
C      FL(2)= SORT(V2/V3)
C      FL(3)=C1*V1/(C2*V1+2.0)
C      FL(5)=V3/C1**2/V1
C      FL(4)=(2.0*C*GAM*V1-C2)/C1
C      U4=C2/2.0
C      BME2=FL(2)**2
C      C5=GAM/C2
C      SMFP=1.0 +C4*BM2
C      SMFP=1.0 +C4*BME2
C      FL(6)=(SMEP/SMFP)**C5*FL(4)
C      FL(7)=SMFP
C      IERR=1
C      RETURN
100     FORMAT (32H EXCEEDED 50 ITERATIONS IN SHOCK)
110     FORMAT (110H ---SHOCK DETACHED IN CLASSICAL WEDGE CALCULATION. UNSHOCK
        1 MODIFIED CHENG-KEMP THEORY IS USED FOR REMAINING ANGLES.)
        END

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SHK 79
UNSHK 80
SHK 81-

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DO 80 JX=JY,N
IXJX=N*(JX-1)+IX
JJX=IXJX+IT
A(IXJX)=A(IXJX)-(A(IXJ)*A(JJX))
B(IX)=B(IX)-(B(J)*A(IXJ))
      BACK SOLUTION
NY=N-1
IT=N*N
DO 110 J=1,NY
IA=IT-J
IB=N-J
IC=N
DO 110 K=1,J
B(IB)=B(IB)-A(IA)*B(IC)
IA=IA-N
IC=IC-1
RETURN
END

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SIM 129
SIM 130-

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SUBROUTINE STUNTS
REAL ACOM(30),ELEMENT(10),HP(20)
DOUBLE PRECISION AA,CAPX,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON C, CARB, CH, CSTA, CTT, CTR, CXTAP, CXB, CXMAX,DATEST,DBTEST,DELTI,
1 CRRB, CRS, CSTA, CTT, CTR, CXTAP, CXB, CXMAX,DATEST,DBTEST,DELTI,
2 CTP, CTPL, DLOGA, DLOGR, DLOGR, DLOGR, DLOGR, DLOGR, DLOGR, DLOGR, DLOGR,
3 DELT2,DELT, DLOGA, DLOGR, DLOGR, DLOGR, DLOGR, DLOGR, DLOGR, DLOGR,
4 SPCTEST, PRES, PRESA, RHOC, RHOP, RHPL, RHTH,ROBARA,ROBARP, SCRP,
5 RHOBAR, RHOC, RHOP, RHPL, RHTH,ROBARA,ROBARP, SCRP,
6 SFN, SHPG, SHC, SL, SL64, SM, SU, SU2, SUMG,
7 TEST, TESTB,TPRINT, TSTOP, UP, ZPA
8 TEST, TESTB,TPRINT, TSTOP, UP, ZPA
COMMON BE(64), BET(20), BLBK(31), CAI(64),CAPXTH(20),
1 CCPJ(20), CEACT(64), CGI(20), CGMU(20), ETAI(64), ETAJ(20),
2 CLNWC(64), CLNPI(64), CMW(20), EGT(20), ETG(20), ETG(20),
3 GJR(20), PERTGJ(20), PGJ(20), PI(64), PICH(64), QM(20),
4 OQ(64), SAJ(20), SDCHI(64), SENT(20), SHJ(20), SHJA(20),
5 SKIL(20), SS(20), TR(30), TRF(20), THEV(20), XMJAT(20),
6 TFD(20), TFE(20), TFK(20), TGM(20), ELJ(10,20), GELJ(10,20),
COMMON BETA(64,20), XNUIJP(64,20)
1 XNUIJ(64,20),XNUIJP(64,20)
COMMON IC, INEQ, INEQV, IP, IRUN, ISC, ISCP1,
1 ISMC, ISMCNR, ISR, ISS, ISSNR, ISSP1, ISSP2, ISSP3, ISSP4,
2 ISW1A, ISW1B, ISW2A, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISW5A,
3 ISW5B, ISW6A, ISW6B, IUPTD, IZFRD, IZFRD, IZFRD, IZFRD, IZFRD,
4 NIT, NNN, NNS, NQS, NQT, NTEST
COMMON IGJ(20), IGM(20), ITB(5), KUR(64,20), LP1J(20,10)
COMMON ACOM,ELEMENT,HP
COMMON /SPEC/ SPRP(43,30)
COMMON /TEMPRY/ SAVEC(20)
COMMON /TRANS1/ 1,0(3,20,20),ZM2(20)
COMMON /TRANS2/ KQ(100),NMQ(100),ISEQ(100),NNKQ,IIIII(50)
COMMON /TRANS3/ VV(5,100)
COMMON /TRANS4/ JQ(5,100)
COMMON /TRANS5/ IIQ(5,100)
COMMON /TRANS6/ V(400),KQ(100),NQ(100),IQ(400),JQ(400),NKQ
COMMON /TRANS7/ NV(15),NN
DATA ASTRK/'*****'/
DIMENSION DELTT(6),NDT(6),TT(74),TPROP(74,4,2)
DATA NDT /10,10,14,10,15,15/, DELTT /100,200,500,1000,2000,
1 5000,
EQUIVALENCE (TT(1),BETA(1,1)), (TPROP(1,1,1),BETA(1,3))
DO 110 J=1,ISS
HQJ=SHJA(J)*CRP/1000
LIM1=1
WRITE (6,290) HP(J),H00
IF (IGJ(J)) 20,10,20
WRITE (6,300) (ASTRSK,K=1,6)
LIM1=1
GO TO 50
WRITE (6,310) (ASTRSK,K=1,12)
LIM2=2
CTMXX=1000,
IF (ETAJ(J),GT,0.) GO TO 50

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LIM1=2
CTMX=-1000.
DO 40 K=1,4
DO 30 I=1,74
TPROP(I,K,I)=0.
CONTINUE
DO 50 M=LIM1,LIM
  TV=0
  II=1
  TV=0
  DO 70 I=1,74
  IT=IT+1
  IZ=I
  IF (IT.LE.NDT(II)) GO TO 60
  II=II+1
  IT=I
  TV=TV+DELTT(II)
  IF (M.EQ.1) TT(I)=TV
  CT=TV/CTAP
  CALL THERM
  TPROP(I,1,M)=XMJAT(J)-SHJA(J)/CT
  TPROP(I,2,M)=(SHJ(J)-SHJA(J))*CRP/1000.
  TP P(I,3,M)=CCPJ(J)*CRA
  I OP(I,4,M)=SFNT(J)*CRA
  IF (CT-I.) 70,90,80
  CONTINUE
CTMX=-1000.
CONTINUE
DO 100 I=1,IZ
WRITE (6,320) TT(I),((TPROP(I,K,M),K=1,4),M=1,LIM)
CONTINUE
RETURN
ENTRY STUNT2
WRITE (6,330)
WRITE (6,340)
WRITE (6,350)
DO 150 L=1,NKQ
  I=ISEQ(L)
  NNQI=NNQ(I)
  DO 140 J=1,NNQI
  IIQJ=IIQ(J,I)
  JJQJ=JJQ(J,I)
  IF (J-I) 120,120,130
  WRITE (6,360) L,I,KKQ(I),(VV(K,I),K=1,5),SPRP(1,IIQJ),SPRP(1,JJQJ)
  GO TO 140
  WRITE (6,370) SPRP(1,IIQJ),SPRP(1,JJQJ)
  CONTINUE
  WRITE (6,330)
  WRITE (6,390)
  WRITE (6,350)
  MV2=1
  LQ2=0
  DO 190 L=1,NKQ
  KK=KKQ(L)

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MV1=MV2+1
MV2=MV2+NV(KK)
LQ1=LQ2+1
LQ2=NO(L)
WRITE (6,390) L, KK, (V(I), I=MV1, MV2)
IF (LQ2.LT.LQ1) GO TO 190
DO 190 LQ=LQ1, LQ2
I=IQ(LQ)
JFJQ(LQ)
IF (LG-LQ1) 160, 160, 170
WRITE (6,400) HP(I), HP(J)
GO TO 190
WRITE (6,370) HP(I), HP(J)
CONTINUE
CONTINUE
IF (ISWB.GT.0) GO TO 240
WRITE (6,250)
ICARD=0
DO 200 I=1, ISS
SAVEC(I)=GJA(I)*CMA
DO 230 I=1, ISS
DO 220 J=I, ISS
WRITE (6,260)
DO 210 K=1, 30
TV=1000*K
IF (TV.GT.1.00001*CTAP) GO TO 220
ICARD=ICARD+1
CALL TRANSP (TV, PRESA)
WRITE (6,270) TV, (G(L, I, J), L=1, 3), I, J, HP(I), HP(J), ICARD
IF (ISW19.NE.-1) GO TO 210
PUNCH 270, TV, (G(L, I, J), L=1, 3), I, J, HP(I), HP(J), ICARD
CONTINUE
CONTINUE
CONTINUE
WRITE (6,280)
RETURN

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C
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C
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FORMAT (I11, 38X, 56H AVERAGED PAIR CROSS-SECTIONS AS FUNCTIONS OF TEMPERATURE)
FORMAT (//2X, 5HTEMP, 8X, 4HQ(1), 8X, 4HQ(2), 8X, 4HQ(3), 5X, 7HINDICES, 5X)
1X, 7HSPECIES, 8X, 5HCOUNT)
FORMAT (F7.0, 1P3E12, 3, 5X, 13, 1H-, 13, 3X, /4, 1H-, A4, 5X, I8)
FORMAT (//10PH NOTE - CHARGED SPECIES CROSS SECTIONS AT ALL TEMPERATURES ARE BASED ON ELECTRON MOLE FRACTION IN RESERVOIR.)
FORMAT (I11, 40X, 22HTHERMAL PROPERTIES OF , A4, 6H (H00=.F10.3, 11H KCSTU 157
1AL/MOLE))
FORMAT (I11, 10X, 3A6, 18H**PHYSICAL MODEL**, 3A6/1H0, 8X, 1HT, 2X, 12F (MUSTU 159
10-H00)/RT, 9X, 5HH-H00, 12X, 2HCP, 12X, 2HS0/29X, 9HKCAL/MOLE, 2X, 12HCAL/MSTU 160
20LE-DEG, 2X, 12HCAL/MOLE-DEG)
FORMAT (I11, 11X, 3A6, 18H**PHYSICAL MODEL**, 3A6, 12X, 3A6, 18H**THERMSTU 161
10 FIT***, 3A6/1H0, 8X, 1HT, 2X, 12H(MUC-H00)/RT, 9X, 5HH-H00, 12X, 2HCP, 12X, 2HS0/29X, 9HKCALSTU 162
2X, 2HS0, 12X, 12H(MUC-H00)/RT, 9X, 5HH-H00, 12X, 2HCP, 12X, 2HS0/29X, 9HKCALSTU 164
3/MOLE, 2X, 12HCAL/MOLF-DEG, 2X, 12HCAL/MOLF-DEG, 29X, 9HKCAL/MOLE, 2X, 12HSTU 165

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4CAL/MOLE-DEG.2X.12HCAL/MOLE-DEG)
 FORMAT (F10.0,4F14.3,1CX,4F14.3)
 FORMAT (1H1.4RX.28HTRANSPORT CROSS SECTION DATA)
 FORMAT (60X.7H(INPUT))
 FORMAT (1HC.5X.4HSTEP.5X.SHINDEX.4X.6HOPTION.1CX.5HVV(1),10X.5HVV(2),10X.5HVV(3),10X.5HVV(4),10X.5HVV(5),9X.1HINTERACTION)
 FORMAT (2I10.1P5E15.3.9X.A4.3H -.A4)
 FORMAT (114X.A4.3H -.A4)
 FORMAT (59X.8H(EDITED))
 FORMAT (110.120.1P5E15.3)
 FORMAT (1.1+.113X.A4.3H -.A4)
 END

STU 166
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SUBROUTINE THERM
SPECIES THERMAL PROPERTIES
LOGICAL MIX,F1
REAL ACOM(30),ELEMENT(10),HP(20)
DOUBLE PRECISION AA,AAA,CAPX,GJ,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A, AFNTS, AFNX, AMACH,
C, CARB, CH, CHA, CLNT,
1 CRRB, CRS, CSTA, CT, CTAB, CLNT,
2 CTP, CTPL, CTT, CXB, CXB,
3 DELT2, DELTAX, DLOGA, DLOGR, DT,
SPCTEST, PRES, PRESA, PRESTH, PRHC, RHAP,
GRHOBAR, RHDC, RHCP, RHPL, RHTH, ROBARA, ROBARP,
7 SEN, SHPG, SC, SL, SL64,
8 TEST, TESTR, TPRINT, TSTOP, TUP,
COMMON RE(64), BLBK(31),
1 CCPJ(20), CEACT(64), CGI(20), CGMU(20),
2 CLNINC(64), CLNPI(64), CMW(20), ETAI(64),
3 GJR(20), PERTGJ(20), PGJ(20),
4 QQ(64), SAJ(20), SDCHI(64), SENT(20),
5 SKIL(20), SS(20), TBJ(30), TFA(20),
6 TFD(20), TFF(20), TFK(20), THEV(20),
COMMON RETA(64,20), ELJ(10,20),
1 XNUIJ(64,20), XNUIJP(64,20)
COMMON IC, IM, INEQ, INEQV, IP, IRUN, ISC, ISCP1,
1 ISMC, ISMCNR, ISR, ISS, ISSNR, ISSP1, ISSP2, ISSP3, ISSP4,
2 ISW1A, ISW1B, ISW2A, ISW2B, ISW3A, ISW3B, ISW4A, ISW4B, ISWSA,
3 ISW5B, ISW6A, ISW6B, IUPD, IZERO, JJK, LC, MI, NFIT,
4 NIT, NNN, NNS, NQS, NQT, NTEST
COMMON IGJ(20), IGM(20), ITR(5), KUR(64,20), LPIJ(20,10)
COMMON ACOM,ELEMENT,HP
DIMENSION AAA(22,24),BTA(64,20),CAPO(31),CCI(20),DGJ(20),GJ(20),SB
J(20),SDGJ(20),SHJAP(20),THEVP(20)
EQUIVALENCE (AA(1,1),AAA(1,1)),(BETA(1,1),BTA(1,1)),
1 (BLBK(1),CAPO(1)),(CAPXTH(1),CCI(1)),(GJA(1),DGJ(1)),
2 (CAPX(1),GJ(1)),(SAJ(1),SBJ(1)),(SS(1),SDGJ(1)),
3 (SHJA(1),SH_P(1)),(THEV(1),THEVP(1))
COMMON /POLYAT/ THEVE(4,20)
COMPUTE ALL THERMAL PROPERTIES
E1=FALSE,
GO TO 10
ENTRY THERM1
COMPUTE SPECIFIC HEATS ONLY
E1=TRUE,
CLGT=ALOG(CT)
CTP=CT*CTAP
CLNT=ALOG(CTP)
DO 320 J=1,ISS
SZ1=0,
SZ2=0,
IF (ETAJ(J),LE,0.) GO TO 30
IF (TGJ(J)) 20,30,20
ZT=CTAP*(CT-CTMXX)
IF (ABS(ZT),GE,500.) GO TO 40
IF (CTMXX,LE,0.) GO TO 40

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MIX=.TRUE.
WTF=C.001*(ZT+500.)
WTHO=1.-WTF
GO TO 50
MIX=.FALSE.
IF (ETAJ(J)) 50,50,70
MIX=.FALSE.
IF (CT-CTMXX) 70,50,50
IF (E1) GO TO 60
THERMO FIT
XMJAT(J)=TFA(J)*(1.-CLNT)-TFK(J)-CT*(TFB(J)+CT*(TFC(J)/2.+CT*(TFD(
1J)/3.+CT*TFE(J)/4.)))+SHJA(J)/CT
SHJ(I)=SHJA(J)+CT*(TFA(J)+CT*(TFB(J)+CT*(TFC(J)+CT*(TFD(J)+CT*TFE(
1J))))
CCPJ(J)=TFA(J)+CT*(2.0*TFB(J)+CT*(3.0*TFC(J)+CT*(4.0*TFD(J)+CT*(5.
1)*TFE(J))))
IF (.NOT.MIX) GO TO 300
XMTF=XMJAT(J)
SHTF=SHJ(J)
CCPTF=CCPJ(J)
STF=SHTF/CT-XMTF
PHYSICAL MODEL
CCPJ(J)=0.
IF (E1) GO TO 80
SHJ(J)=0.
XMJAT(J)=0.
IF (ETAJ(J).LT.3.) GO TO 150
TRANS LATIONAL, ROTATIONAL, AND VIBRATIONAL CONTRIBUTIONS FOR
LINEAR TRIATOMIC SPECIES
Z1=3.5*CLNT
DC 140 K=1,4
IF (INEGV) 90,100,90
Z3=THEVE(K,J)
GO TO 110
Z3=THEVE(K,J)/CT
IF (Z3.GT.88.) GO TO 140
Z4=EXP(Z3)
Z5=Z4-1.
Z5=THEVE(K,J)/Z5
IF (INEQV.NE.IZERO) GO TO 120
Z7=Z3/Z5
GO TO 130
Z7=0.
SZ2=SZ2+Z6/CT
Z8=ALOG(Z4/Z5)
Z9=Z7*(Z4*Z7)
CCPJ(J)=CCPJ(J)+Z9
IF (E1) GO TO 140
XMJAT(J)=XMJAT(J)-Z8
SHJ(J)=SHJ(J)+Z6
CONTINUE

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CCPJ(J)=CCPJ(J)+3.5
IF (E1) GO TO 240
SZ1=SHJ(J)
X MJAT(J)=X MJAT(J)-SAJ(J)-Z1+SHJA(J)/CT+(1.5+ETAJ(J))*CTPL
SHJ(J)=SHJ(J)+3.5*CT+SHJA(J)
GO TO 240
TRANSLATIONAL, ROTATIONAL, AND VIBRATIONAL CONTRIBUTIONS FOR
MONATOMIC AND DIATOMIC SPECIES
150 C Z1=ETAJ(J)-1.0
Z2=Z1+2.5
IF (E1) GO TO 160
X MJAT(J)=0.
160 C CCPJ(J)=0.
IF (INEQV) 170,180,170
Z3=THEV(J)
GO TO 190
180 C Z3=THEV(J)/CT
IF (Z3.GT.88.) GO TO 230
190 C Z4=EXP(Z3)
Z5=Z4-1.
IF (Z1) 200,230,200
Z6=THEV(J)/Z5
IF (INEQV.NE.ZERO) GO TO 210
200 C Z7=Z3/Z5
GO TO 220
210 C Z7=0.0
Z8=ALOG(Z4/Z5)
Z9=Z7*(Z4*Z7)
220 C CCPJ(J)=Z1*Z9
IF (E1) GO TO 230
X MJAT(J)=-(Z1*Z8)
230 C CCPJ(J)=CCPJ(J)+Z2
IF (E1) GO TO 240
SHJ(J)=Z2*CT+SHJA(J)+Z1*Z6
X MJAT(J)=X MJAT(J)+SHJA(J)/CT-(SAJ(J)+Z2*CLGT)
240 C IF (IGM(J).EQ.0) GO TO 290
ELECTRONIC CONTRIBUTIONS (ALL SPECIES)
S1=0.0
S2=0.0
S3=0.0
N=IGM(J)
DO 250 M=1,N
XX=GELJ(M,J)*EXP(-ELJ(M,J)/CT)
S1=S1+XX
S2=S2+XX
S3=S3+XX
X=-XX*ELJ(M,J)
250 C IF (E1) GO TO 260
X MJAT(J)=X MJAT(J)-ALOG(S1)
SHJ(J)=SHJ(J)+S2/S1
260 C IF (S2.GT.1.0D-37) GO TO 270
XX=0.
GO TO 280
270 C XX=S2*S2

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CCPJ(J)=CCPJ(J)+(SI*S3-MW)/((SI*CT)**2)
IF (INEOV.FO.J) GO TO 300
IF (ETAJ(J)>.2) GO TO 310
IF (E1) GO TO 310
ENTROPY CALCULATION
SENT(J)=Z2+(SHJA(J)+S2/S1)/CT+Z1*Z6-XMJAT(J)
GO TO 310
IF (E1) GO TO 310
SENT(J)=SHJ(J)/CT-XMJAT(J)
IF (INEOV.N.O) SENT(J)=SENT(J)+SZ1-SZ2
IF (.NOT.MIX) GO TO 320
MIX PHYSICAL MODEL AND THERMO FIT RESULTS IN TRANSITION
TEMPERATURE RANGE
CCPJ(J)=WTHO*CCPJ(J)+WTF*CCPTF
IF (E1) GO TO 320
XMJAT(J)=WTHO*XMJAT(J)+WTF*XMJATF
SHJ(J)=WTHO*SHJ(J)+WTF*SHJTF
SENT(J)=WTHO*SENT(J)+WTF*SENTF
CONTINUE
RETURN
END
  
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SUBROUTINE THROAT
REAL ACOM(30),ELEMENT(10),HP(20)
LOGICAL ERR
DOUBLE PRECISION AA,CAPX,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON A,AFNTS,AFNX,AMACH,AR,ARBA,ARBB,BZERO,
C,CARB,CARS,CSTA,CH,CLNT,CLNT,CM,CMAX,CTC,CTMAX,CTMX,
CCTO,CTPL,CTT,CX,CTXH,CXMAX,DATEST,DELT,DELTI,
4 DELT2,DELTAX,DLGGA,DLOGR,DY,DZ,ENT,FLUX,HDELX,PCT,
5 PCTEST,PRES,PRESA,PRESB,PRESTH,PRHC,RHAP,RHO,RHOB,
6 RHODBAR,RHOC,RHOP,RHPL,RHTR,ROBAR,ROBARP,SCPG,SDT,
7 SEN,SHPG,SC,SL,SL64,SU,SU2,SUMG,
8 TEST,TESTR,TPRINT,TSTOP,UP,ZP,ZPA
COMMON BE(64),RET(20),ULRK(31),CAI(64),CAPXTH(20),
1 CCPJ(20),CEACT(64),CGI(20),CGMU(20),CHI(64),CHII(20),
2 CLNIMC(64),CLNPI(64),CMW(20),ETAI(64),ETAJ(20),CHI(20),
3 GJ(20),PERTGJ(20),TGJ(20),PI(64),PICH(64),OM(20),
4 GQ(64),SAJ(20),S,CHI(64),SENT(20),SHJ(20),SHJA(20),
5 SKIL(20),SS(20),TB(30),TFA(20),TFR(20),TFC(20),
6 TFC(20),TFK(20),THEV(20),XM,AT(20),XNUI(64)
COMMON BETA(64,20),GLJ(10,20),GFLJ(10,20),
1 XNUIJ(64,20),XNUIJP(64,20),INEQV,IP,IRUN,ISC,ISCP1,
COMMON IC,IM,INEQV,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
1 ISMC,ISMGNR,ISR,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
2 ISW1A,ISW1B,ISW2A,ISW2B,ISW3A,ISW3B,ISW4A,ISW4B,ISW5A,
3 ISW5B,ISW6A,ISW6B,IUPD,IZERO,JJK,LC,MI,NFIT,
4 NI,NN,NNN,NOS,NOT,NTEST
COMMON ACOM,ELEMENT,HP,IGM(20),ITR(5),KUR(64,20),LPIJ(20,10)
COMMON ZERROR,ERR
COMMON /BL/DELEL(2),BLINT(2),XZERO,TWALL,CPWALL,VISROT,DIAM(2),
1 SW,RO,JDIM,IPCINT
COMMON /APL/ATPI(11,2),PARAM(3,12,2),RTHCM(2),NSECT(2),
1 NSECTU(2),ISHAPE(12,2),NPROFL(2),NPRFLS,NBL
COMMON /NEO/OMOST(2),DDELEL(2)
COMMON /AEGECM/ SORTA,S1,S2
COMMON /THRT/ RSA
DATA ONE /1./
IUPD=
CALL GEOMAR (CX,S1,S2)
IF (ERR) RETURN
IF (ISW39.FO.O) GO TO 10
CALL AGSOLN (AFNX,DELEL,ONE,AG,XX)
IF (ERR) RETURN
GO TO 20
AG=AFNX
RSA=PRC*AG/S1
WRITE (6,30) CX,AMACH,AFNX,DLOGA,S1,S2,RSA,(DELEL(I),I=1,NPRFLS)
RETURN

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2.4HRSA=.E10.3.4X.6HDELEL=.E10.3.5X.E10.3//1
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C SUBROUTINE TRANSP (ITAB,P)
C SUBROUTINE TRANSP HAS GENERAL SUPERVISION OF ALL TRANSPORT
C PROPERTY CALCULATIONS IN NATA CODE.
LOGICAL LGR
REAL ACOM(30),ELEMENT(10),HP(20)
DOUBLE PRECISION AA,CAPX,CDIJ
COMMON AA(22,24),CDIJ(20,10),CAPX(20)
COMMON AZ,AFNTS,AFNX,AMACH,
1 C7,CARR,CH,
2 CRPB,CRS,CSTA,
3 CYP,CTPL,CXY,
4 DELT2,DELTA,DLOGA,DLCGR,
5 CTEST,PRESA,DLOGA,DLCGR,
6 RHOBAR,PHOC,RHOP,
7 SEN,SHDG,SC,
8 TEST,TESTB,TPRINT,TSTOP,
1 COMMON BE(64),
2 CLNIMC(64),CLNPI(64),
3 GUP(20),PERTGJ(20),
4 GJ(64),SAJ(20),SDCHI(64),
5 SKIL(20),SS(20),
6 TED(20),TFE(20),
1 COMMON BETA(64,20),XNUIJP(64,20)
1 COMMON IC,IM,INEQ,INEQV,IP,IRUN,ISC,ISCP1,
2 ISWC,ISWCNR,ISR,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
3 ISWA,ISWB,ISW2A,ISW2B,ISW3A,ISW3B,ISW4A,ISW4B,ISW5A,
4 ISWB,ISW6A,ISW6B,IUPD,IZERO,JJK,LC,MI,NFIT,
5 NIT,NNN,NNS,NQT,NTEST
COMMON IGM(20),ITB(5),KUR(64,20),LPIJ(20,10)
COMMON ACDM,ELEMENT,HP
COMMON /TRPORP/ VISC,PRF,SIGMA,FLEWIS
COMMON /TRANS1/ T,O(3,20,20),ZM2(20)
COMMON /TRANS3/ B(20,2),BR(20,2),A(2),X(20),DH(20)
COMMON /TRANS5/ N,IELEC,IDI,ID2
COMMON /RDR/ ISWB
COMMON /TEMPRY/ SAVEC(20)
COMMON /ROROD/ LEWIS,IAMBIP
COMMON /ERROR/ ERR
DIMENSION ZM1(20),C(20,20),ZK(2),CSAVE(20)
GO TO 30
ENTRY TRANSP OF TRANSP DEFINES SPECIES AND INITIALIZES DATA FOR
TRANSPORT CALCULATIONS.
ENTRY TRANSP
N=ISS IS THE NUMBER OF SPECIES TO BE USED IN THE TRANSPORT
CALCULATIONS.
N=ISS
CALL XSECT
IF (ERR) RETURN
CGI IS THE SPECIES MOLECULAR WEIGHT.
DO 10 I=1,N
ZM1(I)=2.8567*CGI(I)
ZM2(I)=.032064*CGI(I)
DO 20 I=1,N

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20  DO 20 J=I,N
    Q(1,I,J)=ZM1(I)+ZM1(J)
    C(J,I)=(ZM1(I)-ZM1(J))/Q(1,I,J)
    C(I,J)=SORT(ZM1(I)*ZM1(J)/Q(1,I,J))
    RETURN
    ENTRY TRANSP CALCULATES TRANSPORT PROPERTIES FOR TEMPERATURE T.
    PRESSURE P, AND MOLE FRACTIONS X(J),
    NOTE - THE MOLE FRACTIONS ARE NOT NECESSARILY IN EQUILIBRIUM AT
    C    THE TEMPERATURE T.
30  T=TTAB
    SAVE CT AND CCP(J) FOR LATER USE.
    CTSAVE=CT
    CTET/CTAP
    DO 40 J=I,N
    CSAVE(J)=CCPJ(J)
    CALCULATE SPECIES SPECIFIC HEATS AT TEMPERATURE T.
    CALL THERM1
    CPTOT=0.
    CMBAVE=0.
    DO 50 JE1,N
    X(J)=SAVEC(J)
    DH(J)=CCPJ(J)-2.5
    CPTOT=CCPJ(J)*X(J)+CPTOT
    CMSAVE=CGI(J)*X(J)+CMSAVE
    CCPJ(J)=CSAVE(J)
    CT=CTSAVE
    COMPUTE COLLISION CROSS SECTIONS. STORE IN Q ARRAY.
    CALL PUTGIN (X(I)*P)
    COMPUTE MATRIX ELEMENTS FOR TRANSPORT CALCULATIONS. STORE IN Q
    C    ARRAY.
    IF (ISWB,LT,0) RETURN
    DO 60 JE1,N
    DO 60 JE2,N
    Q(1,I,J)=C(1,J)*Q(1,I,J)
    Q(2,I,J)=C(1,J)*Q(2,I,J)
    S=SQRT(T)
    FLEWIS=CPTOT/Q(1,IC1,ID2)
    IF (N.EQ.1) GO TO 60
    DO 70 JE2,N
    Q(3,J,J)=Q(2,J,J)
    JE=J-1
    DO 70 IE1,J1
    Q(3,I,J)=3.4375*Q(1,I,J)-.75*C(I,J)*Q(3,I,J)
    Q(1,J,I)=(1.-C(J,I))*2*(Q(3,I,J)-Q(2,I,J))
    Q(2,J,I)=C(J,I)*(3.75*Q(1,I,J)-Q(3,I,J))
    Q(3,I,I)=Q(3,I,J)+Q(2,J,I)-Q(1,J,I)
    Q(3,I,J)=Q(3,I,J)+Q(2,J,I)-Q(1,J,I)
    Q(2,J,I)=(1.66666667*Q(1,I,J)-Q(2,I,J))/(ZM2(I)+ZM2(J))
    Q(2,I,I)=Q(2,I,I)
    COMPUTE TRANSPORT PROPERTIES. VISC=VISCOSITY, PRF=FROZEN PRANDTL
    C    NUMBER, SIGMA=ELECTRICAL CONDUCTIVITY, FLEWIS=ATOM-MOLECULE LEWIS
    DO 90 IE1,N
    R(I,I)=X(I)*Q(1,I,I)

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B(I,2)=X(I)*O(2,I,I)
CALL ELCOND (SIGMA,SSIG)
CALL KINT (ZKINT)
CALL KANDMU (ZK)
ZKINT=ZK(I)+ZKINT
VISC=S#ZK(2)*1.C-3
FLEWIS=FLEWIS/ZKINT*IAMBIP
PRF=8.3167*CPTOT*ZK(2)/ZKINT/CMSAVE
IF (ISW8.EQ.0) RETURN
PRINT TRANSPORT PROPERTY VALUES ONLY IF ISW8.NE.0.
WRITE (6,100)
WRITE (6,110) T,P,VISC,PRF,SIGMA,FLEWIS,CPTOT,CMSAVE
WRITE (6,110) (X(J),J=1,N)
RETURN

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FORMAT (1H )
FORMAT (1P12E11.3)
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SURROUTINE WEDGE
DOUBLE PRECISION ZETA,CAPGAM,ZPP,ZZPP,Z,AMD,AND,GAMD,FL
LOGICAL WEDGEM,AXISYM,FINAL
COMMON /RDKEGG/ ANGLE(10),RADLE(5),WX1,DWX,WXI(20),TWEDGE,WK,
1 NWX,NANGLE,NRADLE,WEDGEM,AXISYM,(SW9B
COMMON /WCD/ TSF
COMMON /BL/ DELEL(2),BLINT(2),XZERO,TWALL,CPWALL,VISROT,DIAM(2),
1 SW,RC,JDIM,IPCINT
COMMON /TRPRDP/ VISC,PR,SIGMA,FLEWIS
COMMON /OUTPUT/ FVOUT(35),GJMF(20)
EQUIVALENCE (GAMMA,FVOUT(10)), (AM,FVOUT(8)), (VMUINF,FVOUT(14)),
1 (TINF,FVOUT(2)), (RHOINF,FVOUT(5)), (UINF,FVOUT(7)),
2 (PINF,FVOUT(A)), (REPF,FVOUT(12)), (GMW,FVOUT(13))
DIMENSION XW(10),PW(10),QW(10),DELSTW(10),YS(10),ZTA(10),UN(2)
DIMENSION FL(7), APR(10), AST(2), ASTQ(10),CH(10)
DIMENSION GFAC(2),GFAK(2)
DATA UN / .,UN,/, RNAME/5HWEDGE/, ZERO /0./, ANMIN /1.E-4/
DATA PI /3.1415927/, AST / .,.,.,/, ALMIN /0.01/
DATA AE /1./, RGC /8.314E7/, SQTPI /1.7724539/
ITHE=1
IAMAX=100
ISD=IABS(ISW9B)
HC=FVOUT(3)/1.P+0.5*(FVOUT(7)*30.48)**2/4.186E7
HW=CPWALL*TWEDGE
HO AND HW ARE IN CAL/GM
HRATIO=HW/HO
TRATIO=TWEDGE/TINF
SEAM*SORT(0.5*GAMMA)
RURE=RCINF*UINF**2
TREF=TSF*(1.+3.*HRATIO)/6.
CALL TRANSF (TREF,PINF)
CE=VISC/VMUINF*TINF/TREF*0.067197
TAU=0.664+1.73*HRATIO
AE=(GAMMA+1.)/2.
A2=AA*
A4=AP*
A5=AA*
A7=AP*
G2=GAMMA*GAMMA
G3=GAMMA*G2
WK2=WK*WK
WK3=WK2*WK
WK4=WK2*WK
EPS=(GAMMA-1.)/(GAMMA+1.)
EPS2=EPS*EPS
OF=WC1*0.08886*PINF*SOPT(RGC*TINF/(2.*PI*GMW))*AE
OFWC2=S**2+(GAMMA-0.5*(GAMMA+1.))*TRATIO/(GAMMA-1.)
/FAC=16.*A7/G3*EPS2/WK4
GFCE=WK/(2.*A2*EPS)
AV2=AA*
DFACE=AE/GAMMA*4.*EPS2*AM2/WK2
CFACE=G2/A5*WK3/R./EPS
CHDFACE=0.332*AA*EPS

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C

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Y5FAC=0.125*G2*WK3/A4/EPS
X5=533.*C*AM2/PEPF
X5=12.*C*(EPS*TAU*AM*AM2)**2/REPF
WRITE (6,250) UN(ITH)
WRITE (6,270) XM
IF (ITH.EQ.2) WRITE (6,280) XS
ITHS=ITH
GFAC(1)=GFAC/AM
GFAC(2)=GAMMA*GFC
DO 240 I=1,NRADLE
TN=2.*RADLE( I )
IF (TN.EQ.0.) TN=1.E-6
EN=AV*SQRT(C/(REPF*TN/12.))
TAUN=TAUN*EN
TAUN2=TAUN*TAUN
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 WED 156
 WED 157
 WED 158
 WED 159
 WED 160
 WED 161
 WED 162
 WED 163
 WED 164
 WED 165

```

60  NW2=1
    ITAB=1
    IT=IT+1
    IF (NW1.GT.NWX) GO TO 70
    WXV1=WXI+(NW1-1)*DWX
    GO TO 80
70  WXV1=1.E30
    IF (NW2.GT.20) GO TO 90
80  WXV2=WXI(NW2)
    GO TO 100
90  WXV2=1.E30
    IF (WXV1-WXV2) 110,110,120
100 IF (WXV1.GT.1.E19) GO TO 130
110 XW(IT)=WXV1
    NW1=NW1+1
    GO TO 140
120 IF (WXV2.GT.1.E19) GO TO 130
    XW(IT)=WXV2
    NW2=NW2+1
    GO TO 140
130 IF (IT.LE.1) GO TO 230
    IT=IT-1
    FINAL=.TRUE.
    GO TO 120
140 XC=XW(IT)*COS(ALPHA)
    ZETA=ZTA*XC
    ZTA(ZTA)=ZETA
    CALL RESOLN (ZETA,CAPGAM,ZZP,ZZPP,Z)
    ZZPS=ZP
    IF (I.H.EQ.1) GO TO 150
    CHENG=KEMP THEORY
    YS(IT)=YSEAK*Z
    P*(IT)=PINF*PFAK*ZZPP
    SQ=SQRT(ZZPS)
    DELSTW(IT)=DFAK*SQ1/ZZPP
    GO TO 150
    C  MODIFIED CHENG-KEMP THEORY
    SQ=SQRT(ZZPS+CMEGA*7IA(IT))
    DELSTW(IT)=DFAK*SQ/(ZZPP+OMEGA)
    PRATIO=PFAK*ZZPP+1.
    P*(IT)=PINF*PRATIO
    YS(IT)=YSEAK*(1.+SQ1)/(ZZPP+OMEGA)+A*XC*TAN(ALPHA)
    CH(IT)=CHD/DELSTW(IT)
    Q*(IT)=CH(IT)*PHOINF*UINF*(HO-HW)*1.8
    ASTQ(IT)=AST(1)
    IF (I.H.EQ.2) GO TO 170
    IF (Q*(IT).L1.CFM(IA)) GO TO 170
    IF M=1
    Q*(IT)=CFM(IA)
    ASTQ(IT)=AST(2)
    IF (IT.LI.10) GO TO 60
170 IF (ITAB.EQ.1) GO TO 190
180 WRITE (5,290)
    GO TO 200
190 CPGM=CAPGAM
  
```

```

200 WRITE (6,300) RADLE(IR),ANGLE(IA),CPGM, EGA
    WRITE (6,310) (XW(J),J=1,IT)
    WRITE (6,320) (PWW(J),J=1,IT)
    WRITE (6,330) (CWW(J),ASTO(J),J=1,IT)
    WRITE (6,340) (CH(J),J=1,IT)
    WRITE (6,350) (DELSTW(J),J=1,IT)
    IF (IS.EQ.0) GO TO 220
    IF (IS.EQ.2) GO TO 210
    WRITE (6,360) (YS(J),J=1,IT)
    IF (IS9.EQ.1) GO TO 220
    WRITE (6,370) (ZTA(J),J=1,IT)
    IF (IP.NE.C) WRITE (6,380)
    IF (FINAL) GO TO 230
    IT=C
    ITAB=ITAB+1
    GO TO 60
230 CONTINUE
    ITHS=ITHS
    CCNTINUE
    IF (ITH.EQ.2.CR.IS9.EQ.ISW9B) GO TO 250
    ITH=2
    GO TO 10
250 RETURN
C
C
C
260 FORMAT (1HC,43X,46H*****CONDITIONS ON WEDGE MODELS*****//WED 192
152X,A2,26HMODIFIED CHENG-KEMP THEORY)
270 FORMAT (45H MERGING EFFECTS SIGNIFICANT FOR XW LESS THAN.1PE10.2,8WED 194
1H INCHES.)
280 FORMAT (45H STRONG-INTERACTION APPROXIMATION BREAKS DOWN FOR XW GRWED 196
1EATER THAN,1PE10.2,8H INCHES.)
290 FORMAT (1H0)
300 FORMAT (31HC*****LEADING-EDGE RADIUS =.F6.3,25H INCH. ANGLE OWED 199
1E ATTACK =.F5.1,26H DEGREES. CAPITAL GAMMA =.1PE10.2,9H. OMEGA =,WED 200
2E10.2,9H*****//WED 201
310 FORMAT (12H XW (INCHES),9X,1H=,1P10E11.2)
320 FORMAT (10H PWW (ATM),11X,1H=,1P10E11.2)
330 FORMAT (22H QWW (BTU/50 FT-SEC) =.1X,10(1PE10.2,A1))
340 FORMAT (15H STANTON NUMBER,6X,1H=,1P10E11.2)
350 FORMAT (14H DELSTW (INCH),7X,1H=,1P10E11.2)
360 FORMAT (12H YS (INCHES),9X,1H=,1P10E11.2)
370 FORMAT (5H ZETA,16X,1H=,1P10E11.2)
380 FORMAT (5X,22H* FREE-MOLECULE LIMIT)
    END
WED 166
WED 167
WED 168
WED 169
WED 170
WED 171
WED 172
WED 173
WED 174
WED 175
WED 176
WED 177
WED 178
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WED 199
WED 200
WED 201
WED 202
WED 203
WED 204
WED 205
WED 206
WED 207
WED 208
WED 209
WED 210-

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1  SUBROUTINE WESOLN (ZETA,CAPGAM,ZZP,ZZPF,
2  SOLUTION OF THE CHENG EQUATION FOR FLOW OVER A WEDGE
3  IMPLICIT REAL*8 (A-H,O-Z)
4  KEEP THIS IMPLICIT STATEMENT IN THE UNIVAC 1103 VERSION
5  LOGICAL CALLED
6  DATA R0,31/1.81712059D0,1.732059800/
7  NAMELIST/REFDMP/XL,SQTL,Q1,Q2,Q3,C1,Q5,ZT, DLDZT
8  IF (CALLED) GO TO 10
9  CALLED=.TRUE.
10 C1=1.D0
11 C3=1.D0/3.D0
12 C22=22.D0/9.D0
13 C9=9.D0
14 C463=46.D0/3.D0
15 C103=10.D0/3.D0
16 C4=4.D0
17 C2=1.D0/500
18 C2=2.D0
19 C769=76.D0/9.D0
20 P1=ZETA**C3
21 P2=DSORT(ZETA)
22 IF (ZETA.GT.1.D-8) GO TO 20
23 HODGER'S ANALYTICAL SOLUTION FOR SMALL ZETA
24 P12=P1**2
25 Z1=.05096*P12+C,50869*P1*P2-0.0249*ZETA
26 XL=1.81711*P1+1.25975*P2+C.14712*P12
27 DLDZT=3.6057/P12+C.62988/P2+C.09808/P1
28 GO TO 30
29 ITERATIVE SOLUTION BASED ON CHENG/KEMP EXACT SOLUTION FOR GAMMA=0.
30 XLE=3*P1+P1*P2
31 N=0
32 IF (N.GT.20) GO TO 40
33 SQTL=DSORT(XL)
34 Q1=Q1+SQTL
35 Q2=Q1*Q1
36 Q3=Q1**2
37 Q+=Q2*Q2
38 Q5=CLOG(Q1)
39 ZT=C3*Q4-C229*Q3+C9*Q2-C463*Q1+Q5*(C103-C4*SQTL+C2*Q5)+C769
40 ZEC2*(SQTL-XL/C2+C3*XL**C32-Q5)
41 DLDZT=Q1/Z
42 IF (CABS(ZT-ZETA)/ZETA.LE.1.D-4) GO TO 50
43 XLE=XL-(ZT-ZETA)*DLDZT
44 N=N+1
45 GO TO 30
46 WRITE (6,60) ZETA,CAPGAM
47 WRITE (6,REFDMP)
48 C950=CAPGAM**2
49 ZZP=XL+CG50*ZETA
50 ZZPF=DLDZT+CG50
51 ZPDSORT(Z**2+(CAPGAM*ZETA)**2)
52 RETURN
53
54
55

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WES 56
WES 57-

7HCAPGAM=E15.8)
END

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SUBROUTINE XSECT INPUT CROSS SECTION DATA FOR USE IN
 SUBROUTINE XSECT EDITS
 TRANSPORT CALCULATIONS.

LOGICAL ERR
 REAL ACCM(30),ELMENT(10),HP(20)
 DOUBLE PRECISION AA,CAPX,COIJ
 COMMON AA(22,24),COIJ(20,10),CAPX(20)
 COMMON A,AFNTS,AFNX,AMACH,
 C,CH,CST,CTA,CTB,CTAB,AR,ARBA,ARBB,BZERO,
 CRRS,CRS,CTPL,CTP,CTAX,DLOGA,DLOGR,PRESTH,PRH,PRHC,PRHAP,RHO,
 DELT2,DELTA,DLOGA,DLOGR,PRESTH,PRH,PRHC,PRHAP,RHO,
 SPCTEST,PRESA,PRESTH,PRH,PRHC,PRHAP,RHO,
 6RHOBAR,RHOC,RHOP,SC,SL,SL64,SU,SU2,SUMG,
 7SEN,SHPG,SC,SL,SL64,SU,SU2,SUMG,
 8TEST,TESTR,TPRINT,TSTOP,UP,ZPA
 COMMON BE(64),BET(20),BLBK(31),CAI(64),CAPXTH(20),
 1CCPJ(20),CEACT(64),CGI(20),CGMU(20),CHI(64),
 2CLNMC(64),CLNPI(64),CMW(20),ETAI(64),ETAJ(20),GJA(20),
 3GJB(20),PERTGJ(20),PGJ(20),PI(64),PICH(64),QM(20),
 4OG(64),SAJ(20),SDCHI(64),SENT(20),SHJ(20),SHJA(20),
 5SKIL(20),SS(20),TR(30),TFA(20),TFB(20),TFC(20),
 6TFD(20),TFE(20),TFK(20),THEV(20),XNJAT(20),
 COMMON BETA(64,20),ELJ(10,20),GELJ(10,20),
 1XNUIJ(64,20),XNUJJP(64,20)
 COMMON IC,INEQ,INERV,IP,IRUN,ISC,ISCP1,
 1ISVC,ISMENR,ISS,ISSNR,ISSP1,ISSP2,ISSP3,ISSP4,
 2ISW1A,ISW1B,ISW2A,ISW2B,ISW3A,ISW3B,ISW4A,ISW4B,ISWSA,
 3ISW5A,ISW6A,ISW6B,IUPD,IZERO,IJK,LC,MI,NFIT,
 4NIT,NNN,NNS,NQS,NQT,NTEST
 COMMON IGJ(20),IGM(20),ITB(5),KUR(64,20),LPIJ(20,10)
 COMMON ACOM,ELMENT,HP
 DIMENSION VCDUL(2,2,2,2)
 DIMENSION IZ(20)
 COMMON /TRANS1/ KQ(100),NNQ(100),ISEQ(100),NNKQ,1(50)
 COMMON /TRANS2/ KQ(100),NNQ(100),ISEQ(100),NNKQ,1(50)
 COMMON /TRANS3/ IIG(5,100)
 COMMON /TRANS4/ JJO(5,100)
 COMMON /TRANS5/ VV(5,100)
 COMMON /TRANS6/ LSS,IELEC,IDI,1D2
 COMMON /TRANS7/ V(400),KQ(100),NJ(100),IG(400),JQ(400),NKQ
 COMMON /TRANS8/ NV(15),N
 COMMON /LN/IATCM,IMDL,JATOM,JMOL
 COMMON /MASSFL/ SWASS,CTMXXI,TSIOP1,IS(20)
 COMMON /ERROR/ ERR
 NAMELIST /XSDMP/ 1.0,ZM2,1,LSS,IELEC,IDI,1D2,V,KQ,NQ,IQ,JQ,
 1 NQ,NV,N
 DATA VCDUL/0.994,1.71,3.41,2.01,3.41,2.01,16.1,71.994,1.62,2*0.,
 1 3.41,1.20,3.0,0./
 DATA IZ /20*0./
 N=ISS
 IELEC=IS(1)
 IF (IELEC.NE.1) IELEC=0
 FIND CORRESPONDENCE BETWEEN INPUT SPECIES AND CROSS SECTION DATA.
 JJ=50

1 XSE
 2 XSE
 3 XSE
 4 XSE
 5 XSE
 6 XSE
 7 XSE
 8 XSE
 9 XSE
 10 XSE
 11 XSE
 12 XSE
 13 XSE
 14 XSE
 15 XSE
 16 XSE
 17 XSE
 18 XSE
 19 XSE
 20 XSE
 21 XSE
 22 XSE
 23 XSE
 24 XSE
 25 XSE
 26 XSE
 27 XSE
 28 XSE
 29 XSE
 30 XSE
 31 XSE
 32 XSE
 33 XSE
 34 XSE
 35 XSE
 36 XSE
 37 XSE
 38 XSE
 39 XSE
 40 XSE
 41 XSE
 42 XSE
 43 XSE
 44 XSE
 45 XSE
 46 XSE
 47 XSE
 48 XSE
 49 XSE
 50 XSE
 51 XSE
 52 XSE
 53 XSE
 54 XSE
 55 XSE

```

10  DO 30 K=1.50
    J=51-K
    DO 10 L=1.N
    IF (IS(L)-J) 10,20.10
    CONTINUE
    I(J)=JJ
    JJ=JJ-1
    GO TO 30
    I(J)=L
    CONTINUE
    FIND INDICES OF SPECIES TO BE USED IN LEWIS NUMBER CALCULATION.
    ID1=MINO(IATOM,IMOL)
    ID2=MAXO(IATOM,IMOL)
    CALL CXSECT (ID1,ID2)
    RE-INDEX CROSS SECTION DATA TO CORRESPOND TO ORDER OF INPUT
    SPECIES. SET Q(1,L,J)=1.0 FOR ALL PAIRS OF SPECIES FOR WHICH
    CROSS SECTION DATA ARE PROVIDED.
40  DO 50 J=1.N
    DO 50 L=1.N
    Q(1,J,L)=0.
    Q(1,2,1)=1.
    NKQ=MINO(NNKQ,99)
    LQ=0
    V(1)=1.
    KV=1
    KIJEJ
    DO 90 K=1,NKJ
    M=ISEQ(K)
    NK=KQ(M)
    KA(K)=K
    NL=NV(NK)
    IF (NK.EQ.0) GO TO 70
    DO 60 KM=1,NK
    KV=KV+1
    V(KV)=V(KM,N)
    CONTINUE
    NK=NQ(M)
    LQ=LQ+NK
    NQ(K)=LO
    IF (NK.EQ.0) GO TO 90
    DO 80 KM=1,NK
    KIJEKI+1
    LEI:Q(KM,M)
    JEJ:G(KM,N)
    CALL CXSECT (L,J)
    IF (J.LE.N.AND.L.LE.N) Q(1,L,J)=1.
    IQ(KIJE)
    JQ(KIJE)
    CONTINUE
    FOR K=9 OR 12. THE V-ARRAY DEPENDS ON THE INDEXING OF THE
    SPECIES. THESE DATA MUST ALSO BE RE-INDEXED.
    LEQ
    CALL BXSECT (L,MV,LL,J)
    IF (L.EQ.0) GO TO 110
    CALL CXSECT (LL,J)

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111 IF (J.LE.N.AND.LL.LE.N) O(I,LL,J)=1.
112 GO TO 100
113 EDIT CROSS SECTION DATA TO DELETE SPECIES NOT EQUIPPED FOR
114 TRANSPORT CALCULATIONS.
115 LOI=1
116 FOR KQ=9 OR 12, IT MAY BE NECESSARY TO COMPUTE CROSS SECTIONS FOR
117 SPECIES NOT INCLUDED IN TRANSPORT CALCULATIONS. STATEMENTS 120 TO
118 140 CHECKS FOR THIS POSSIBILITY. IF SUCH CROSS SECTIONS ARE FOUND,
119 STATEMENTS 140 TO 300 REVISE THE DATA TO ELIMINATE THEM FROM THE
120 CALCULATIONS. STATEMENTS 300 THROUGH 380 THEN PROCEED TO EDIT THE
121 CROSS SECTION DATA TO REMOVE SPECIES NOT REQUIRED FOR THE
122 TRANSPORT CALCULATIONS.
123 CALL BXSECT (L,MV,LL,J)
124 IF (L.EQ.0) GO TO 300
125 IF (J.LE.N.AND.LL.LE.N) GO TO 120
126 IF (L.GT.1) LOI=NQ(L-1)+1
127 LQ2=NQ(L)
128 IF (LQ1.GT.LQ2) GO TO 120
129 DO 130 LQ=LQ1,LQ2
130 IF (JQ(LQ).LF.N.AND.IQ(LQ).LE.N) GO TO 140
131 CONTINUE
132 GO TO 120
133 CROSS SECTIONS ARE REQUIRED FOR THE PAIR (LL,J). FIND UNUSED
134 STORAGE LOCATION FOR THEM.
135 DO 150 II=2,N
136 II=II-1
137 DO 150 JJ=1,II
138 IF (Q(I,II,JJ).EQ.0.) GO TO 190
139 CONTINUE
140 NO STORAGE SPACE AVAILABLE. ADD AN ADDITIONAL SPECIES TO CROSS
141 SECTION COMPUTATIONS.
142 IF (N.EQ.20) GO TO 540
143 N=N+1
144 JJ=MAX0(J,LL)
145 DO 160 J=1,50
146 IF (I(J).EQ.N) GO TO 170
147 CONTINUE
148 DO 180 K=1,50
149 IF (I(K).NE.JJ) GO TO 180
150 I(J)=JJ
151 I(K)=N
152 GO TO 40
153 CONTINUE
154 GO TO 540
155 STORE CROSS SECTIONS FOR PAIR (LL,J) IN LOCATION (II,JJ).
156 O(I,II,JJ)=1.
157 IT IS FIRST NECESSARY TO CONVERT ANY KQ=10 STEPS FOR THE PAIR TO
158 KQ=12.
159 NSV=NKQ
160 L=0
161 LOI=0
162 L=L+1
163 IF (KQ(L).NE.10) GO TO 270
164 LQ2=NQ(L)
165 IF (L.GT.1) LOI=NQ(L-1)

```

```

210 IF (LQ1,GE,LQ2) GO TO 270
    LQ=LQ1
    LG=LQ+1
    IV=IO(LQ)
    JV=JQ(LQ)
    IF ((LL,NE,J,AND,(IV,NE,LL,OR,JV,NE,J)),OR,(IV,NE,LL,AND,JV,NE,J))
        1 GO TO 260
    C CONVERSION TO KQ=12 MUST BE MADE FOR CURRENT STEP.
    C FIND LOCATION MM AT WHICH NEW DATA ARE TO BE INSERTED IN V-ARRAY.
    LKQ=L
    LV=0
    CALL BXSECT (LV,MM,K,L)
    IF (LV,NE,0) GO TO 220
    REVISE CROSS SECTION DATA
    MV=MMV
    DO 230 K=MM,MMV
        V(MV+4)=V(MV)
        MV=MV-1
        V(MV+2)=IV
        V(MV+3)=V(MV+2)
        V(MV+4)=JV
        V(MV+5)=V(MV+4)
        MMV=MMV+4
        LQ1=LQ1+1
    LV=NSV
    DO 240 K=NKQ,NSV
        KQ(LV+1)=KQ(LV)
        NQ(LV+1)=NQ(LV)
        LV=LV-1
        NSV=NSV+1
        NO(LV+1)=LQ1
        L=LV+2
        LV=LQ
    DO 250 K=LQ1,LQ
        LV=LV-1
        IO(LV+1)=IO(LV)
        JO(LV+1)=JO(LV)
        IO(LQ1)=IV
        JO(LQ1)=JV
        KQ(NKQ)=I2
    CONTINUE
    IF (LQ,LT,LQ2) GO TO 210
    CONTINUE
    IF (L,LT,NSV) GO TO 200
    NKQ=NSV
    C CONVERSION OF KQ=10 TO KQ=12 COMPLETED. NOW REVISE DATA TO CHANGE
    C PAIR (LL,J) TO (II,JJ).
    LQ2=NQ(NKQ)
    DO 280 LQ=1,LQ2
        IF (IO(LQ),NE,LL,OR,JO(LQ),NE,J) GO TO 280
        IO(LQ)=II
        JO(LQ)=JJ
    CONTINUE
    L=0
    CALL BXSECT (L,MV,IV,JV)
220
C
230
240
250
260
270
.C
C
280
290

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E

E

This document contains information which is exempt from public release under Executive Order 12958, Section 1.4, as amended, because its disclosure could result in the identification of a source of information or the disclosure of information which could be injurious to the national defense.

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C      IF (L.EQ.0) GO TO 110
C      IF (IV.NE.LL.OR.JV.NE.J) GO TO 290
C      JV=JJ
C      GO TO 290
C      EDIT CROSS SECTION DATA TO ELIMINATE SPECIES NOT REQUIRED IN
C      COMPUTATIONS.
300    DO 310 LQ=1,400
310    IF (IQ(LQ).GT.N.OR.JQ(LQ).GT.N) IO(LQ)=0
        CONTINUE
        NEWL=0
        NEWLO=0
        NEWMV=1
        MVI=2
        NSV=NKQ
        LQ1=1
        KO(100)=0
        DO 370 L=1,NSV
            K=KO(L)
            LQ2=NQ(L)
            IF (LQ2.LI.LQ1) GO TO 330
            DO 320 LQ=LQ1,LQ2
                IF (IQ(LQ).NE.0) NEWLO=NEWLO+1
            CONTINUE
            LQ1=LQ2+1
            IF (K.EQ.10) GO TO 350
            IF (NEWLO.FQ.NO(NEWL)) GO TO 360
            MV2=MV1+NV(K)-1
            DO 340 MV=MV1,MV2
                NEWMV=NEWMV+1
            V(NEWMV)=V(MV)
            NEWL=NEWL+1
            KO(NEWL)=K
            NO(NEWL)=NEWLO
            GO TO 370
            NKQ=NKQ-1
            MVI=MV1+NV(K)
            LL=0
            DO 380 LQ=1,400
            IF (IQ(LQ).EQ.0) GO TO 380
            LL=LL+1
            IQ(LL)=IQ(LQ)
            JQ(LL)=JQ(LQ)
            CONTINUE
            COMPUTE UNSPECIFIED NEUTRAL-NEUTRAL CROSS SECTIONS FROM EMPIRICAL
            MIXING LAW. COMPUTATIONS ARE PERFORMED AT LAST SPECIFIED STEP
            WITH KQ=10; OR, IF NO STEPS WITH KQ=10 ARE SPECIFIED, AFTER ALL
            SPECIFIED CROSS SECTIONS HAVE BEEN COMPUTED.
            LL=NKQ+1
            DO 390 L=1,NKQ
            IF (KO(L).EQ.10) LL=L
            CONTINUE
            IF (LL.LE.NKQ) GO TO 400
            KO(LL)=10
            NO(LL)=NO(LL-1)

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221 XSE
222 XSE
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232 XSE
233 XSE
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253 XSE
254 XSE
255 XSE
256 XSE
257 XSE
258 XSE
259 XSE
260 XSE
261 XSE
262 XSE
263 XSE
264 XSE
265 XSE
266 XSE
267 XSE
268 XSE
269 XSE
270 XSE
271 XSE
272 XSE
273 XSE
274 XSE
275 XSE

```

400 NKQ=NKQ+1
      NN=0
      DO 420 L=1,ISS
      IF (IELEC.EQ.0) GO TO 410
      IZ(L)=LPIJ(L,1)
      IF (IZ(L).NE.0) GO TO 420
      NN=NN+1
      I(NN)=L
      IF (Q(I,L).EQ.0.) GO TO 540
      CONTINUE
      CALL AXSECT (LL,NN)
      FOR IONIZED SPECIES, USE EFFECTIVE COULOMB CROSS SECTIONS CHOSEN
      TO FIT TRANSPORT CALCULATIONS OF SPITZER.
      IF (LL.FQ.NKQ.AND.NQ(LL).EQ.NQ(LL-1)) GC TO 430
      NKQ=NKQ+1
      NQ(NKQ)=NQ(NKQ-1)
      DO 480 J=1,2
      IE=IELEC
      KK=J
      DO 470 JJ=1,2
      NN=0
      DO 450 L=1,ISS
      II=IABS(IZ(L))
      IF (L.FQ.IE.OR.(II.NE.J.AND.II.NE.KK)) GO TO 450
      NN=NN+1
      I(NN)=L
      CONTINUE
      CALL AXSECT (NKQ,NN)
      IF (NQ(NKQ).EQ.NQ(NKQ-1)) GO TO 460
      KQ(NKQ)=2
      NKQ=NKQ+1
      NQ(NKQ)=NQ(NKQ-1)
      V(NEWMV+1)=VCOUL(1,KK,J,JJ)
      V(NEWMV+2)=VCOUL(2,KK,J,JJ)
      NEWMV=NEWMV+2
      IF (KK.LE.1) GO TO 470
      KK=KK-1
      GO TO 440
      IE=0
      CONTINUE
      NN=0
      DO 490 L=1,ISS
      IF (IZ(L).EQ.0) GO TO 490
      NN=NN+1
      I(NN)=L
      CONTINUE
      CALL AXSECT (NKQ,NN)
      IF (NQ(NKQ).NE.NQ(NKQ-1)) GO TO 540
      SET UNSPECIFIED ION-NEUTRAL CROSS SECTIONS EQUAL TO TYPICAL VALUE
      ESTIMATED FOR N-0+ INTERACTION.
      DO 500 L=1,ISS
      I(L)=L
      CALL AXSECT (NKQ,ISS)
      IF (NQ(NKQ-1).NE.NQ(NKQ)) GO TO 510
      NKQ=NKQ-1

```

```

XSE 276
XSE 277
XSE 278
XSE 279
XSE 280
XSE 281
XSE 282
XSE 283
XSE 284
XSE 285
XSE 286
XSE 287
XSE 288
XSE 289
XSE 290
XSE 291
XSE 292
XSE 293
XSE 294
XSE 295
XSE 296
XSE 297
XSE 298
XSE 299
XSE 300
XSE 301
XSE 302
XSE 303
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XSE 306
XSE 307
XSE 308
XSE 309
XSE 310
XSE 311
XSE 312
XSE 313
XSE 314
XSE 315
XSE 316
XSE 317
XSE 318
XSE 319
XSE 320
XSE 321
XSE 322
XSE 323
XSE 324
XSE 325
XSE 326
XSE 327
XSE 328
XSE 329
XSE 330

```

510 GO TO 520
 K?(NKQ)=6
 V(NEWMV+1)=996.
 V(NEWMV+2)=5.
 CONTINUE
 520 IF (NKQ.GT.100) GO TO 540
 RETURN
 530 INSUFFICIENT CROSS SECTION DATA AVAILABLE. TERMINATE CASE.
 C WRITE (6,550)
 540 ERR=.TRUE.
 C WRITE (6,XSDMP)
 C GO TO 530

550 FORMAT (/116HOTRANSPORT PROPERTIES OF DESIRED MIXTURE CANNOT BE CALCULATED FROM AVAILABLE DATA. REVISE CROSS SECTION INPUT DATA.)
 END

XSE 331
 XSE 332
 XSE 333
 XSE 334
 XSE 335
 XSE 336
 XSE 337
 XSE 338
 XSE 339
 XSE 340
 XSE 341
 XSE 342
 XSE 343
 XSE 344
 XSE 345
 XSE 346
 XSE 347
 XSE 348-

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APPENDIX

PROGRAMMER'S MANUAL FOR THE NOZFIT CODE

The present appendix documents the programming of the NOZFIT code, a small computer program for generating NATA-type curvefits to nozzle profiles. A user's manual for NOZFIT is presented in Appendix D of Volume II of this report (ref. 2).

Section A.1 of this appendix analyzes the NOZFIT main program and the subroutines used to produce plots of nozzle profile curvefits. Section A.2 is a glossary of Fortran symbols for NOZFIT. Finally, a source listing of NOZFIT and its associated subroutines is presented in Section A.3.

A.1 Analysis of NOZFIT Routines

This section discusses and explains the main program and each subroutine of the NOZFIT code. The purpose of this exposition is to provide an entry into the coding for programmers who wish to analyze errors, make corrections, or introduce modifications into the code.

A.1.1 NOZFIT Main Program

NOZFIT consists of two main sections. The first section computes the parameters of the profile fit from the input data (see Appendix D of Volume II); the second produces printed, punched, and plotted output.

(1) Computation of profile fit parameters. -- The parameters* of the curvefit to the nozzle or channel profile are calculated in the loop DØ 140, L = 1, 2. For L = 1, the parameters of the upstream sections are determined; for L = 2, the downstream sections are treated. For each region, the calculations start at the throat and proceed away from the throat. The index K enumerates the profile sections for

*The profile fit parameters and the analytical forms in which they appear are defined in Section 4.3 of Volume I (ref. 1).

each region in the order in which they are determined. The index J labels the sections for the entire nozzle, both upstream and downstream, in sequence starting at the upstream end. Finally, JP indexes the downstream boundaries of the profile sections in sequence starting at the upstream end.

The algorithm used to determine the parameters for each section depends upon the input "condition" index, ICØND(J).

a. ICØND(J) = 1

In each region (upstream or downstream), the first section treated is the one adjacent to the throat. This section is required to be a circular arc convex toward the axis (ISHAPE(J) = 2), and must be specified as a throat section (ICØND(J) = 1). For such a throat section, dy/dx must be zero at x = 0. Thus, equation I(122) shows that the abscissa P₂ of the circle center must be zero.

$$P_2 = 0 \quad (178)$$

The circle radius is equal to PAR(1,J):

$$P_3 = 2.54 \cdot \text{PAR}(1,J) \quad (179)$$

where the numerical factor converts from the inch unit used in the input to the centimeter unit used in the output. Finally, the ordinate of the circle center is equal to the throat radius plus the circle radius (see figure 28):

$$P_1 = P_3 + \frac{1}{2} \cdot 2.54 \cdot \text{DTH} \quad (180)$$

where DTH is the throat diameter in inches.

b. ICØND(J) = 2

For the condition ICØND(J) = 2, the current profile section is assumed to be straight (ISHAPE(J) = 1) and tangent to an adjacent circular section nearer the throat. The index of the section nearer the throat is denoted by INDEX. The profile parameters for this section are already known, because of the order which the sections are treated.

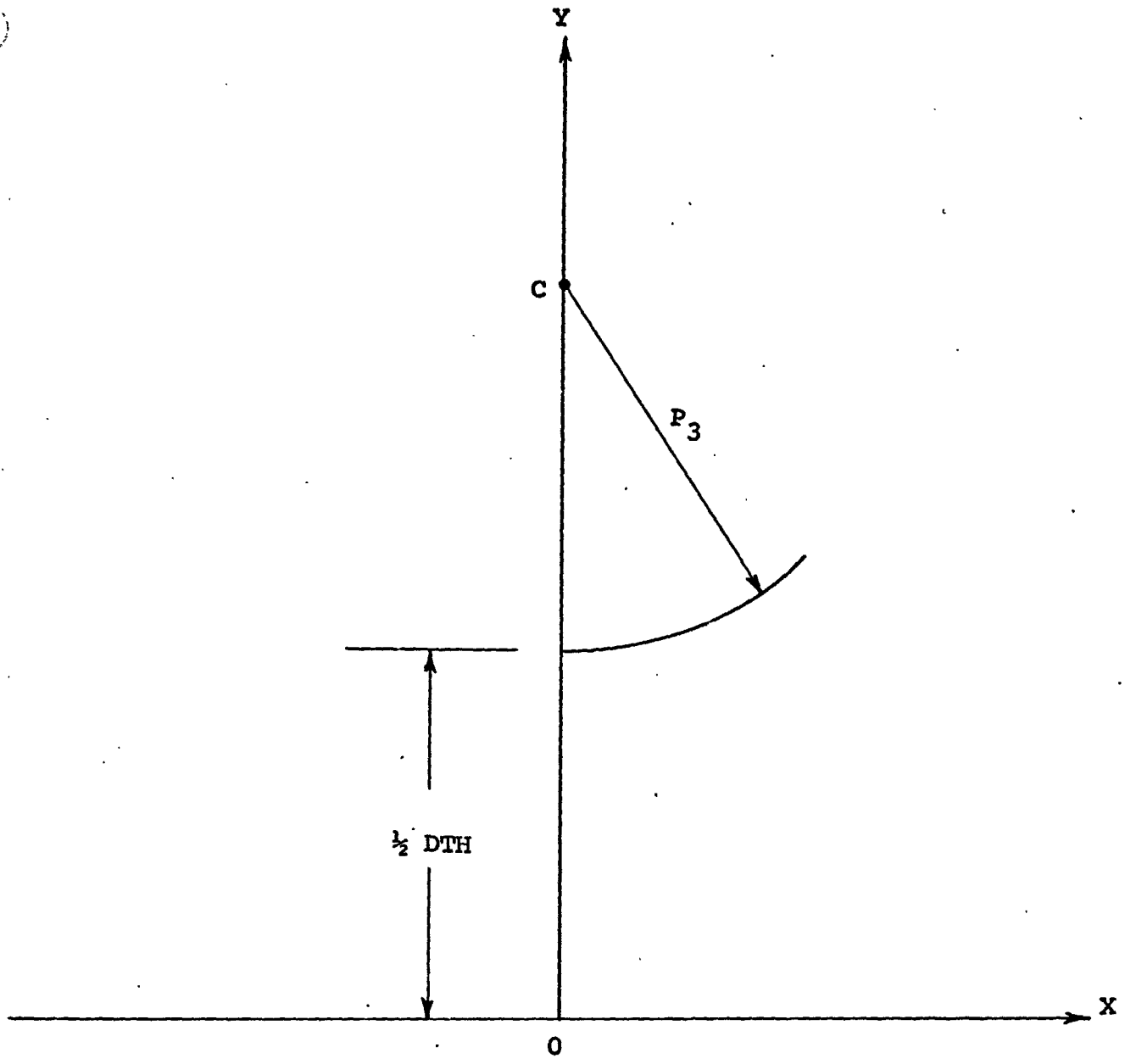


Figure 28. Geometry of the Condition $IC\emptyset ND(J) = 1$

Since the sections can be either upstream or downstream of the throat, and since the adjacent section INDEX can be either convex (ISHAPE(INDEX) = 2) or concave (ISHAPE(INDEX) = 3) toward the axis, there are four cases to be considered, as illustrated in figure 29. These cases are all treated in one set of formulas with the aid of the following symbols:

$$S_1 = \begin{cases} -1 & \text{upstream} \\ +1 & \text{downstream} \end{cases} \quad (181)$$

$$S_2 = \begin{cases} -1 & \text{for circle bottom, ISHAPE(INDEX) = 2} \\ +1 & \text{for circle top, ISHAPE(INDEX) = 3} \end{cases} \quad (182)$$

The symbol S_1 has the same sign as the slope of the current, straight section; S_2 is negative if the circle center C is above the line passing through the straight section, positive if the center is below.

The point of tangency of the straight and circular sections is marked by a "1" in each part of figure 29. The coordinates of this point are

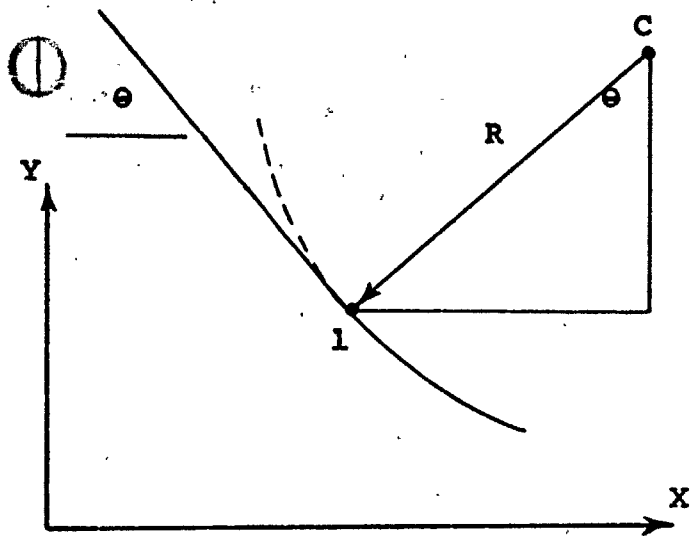
$$X_1 = X_C - S_1 S_2 R \sin \theta \quad (183a)$$

$$Y_1 = Y_C + S_2 R \cos \theta \quad (183b)$$

Here $R = \text{PARAM}(3, \text{INDEX})$ is the circle radius, and $\theta = \text{PAR}(1, J)$ is the angle of inclination of the straight section to the nozzle axis, taken positive. It is easy to show that the internal angle of the triangle at the vertex C is equal to θ . Also X_C and Y_C are the coordinates of the circle center in the adjacent circular section nearer the throat:

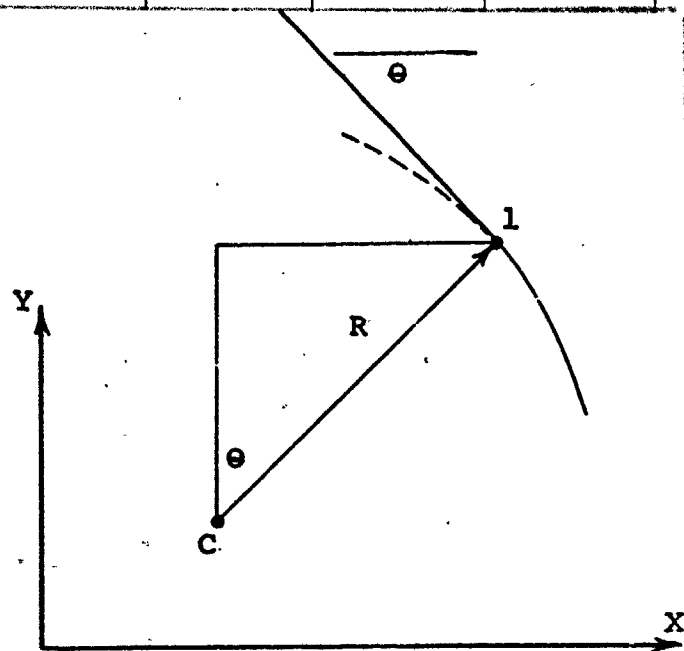
$$X_C = \text{PARAM}(2, \text{INDEX}) \quad (183c)$$

$$Y_C = \text{PARAM}(1, \text{INDEX}) \quad (183d)$$



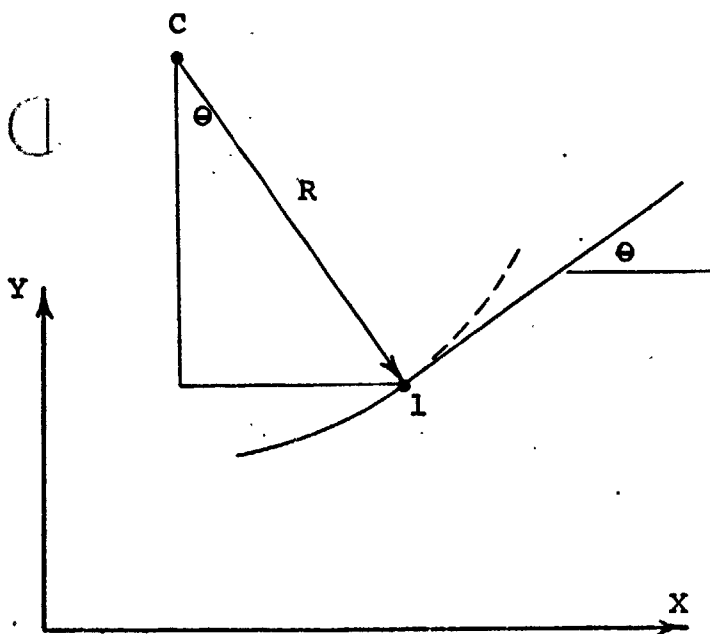
(a)

Upstream, ISHAPE(INDEX) = 2



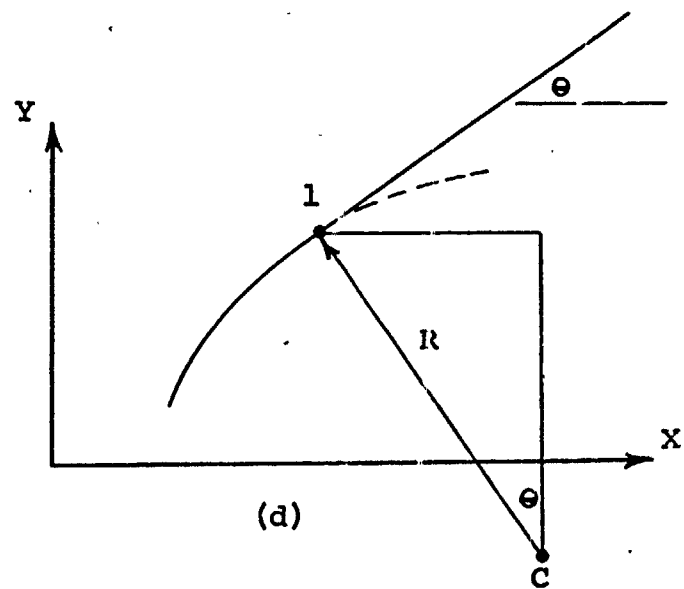
(b)

Upstream, ISHAPE(INDEX) = 3



(c)

Downstream, ISHAPE(INDEX) = 2



(d)

Downstream, ISHAPE(INDEX) = 3

Figure 29. Geometry of the Condition ICND(J) = 2

The equation for the straight section is

$$Y - Y_1 = S_1 \tan \theta \cdot (X - X_1) \quad (184)$$

Thus, from equation I(121),

$$P_1 = Y_1 - S_1 X_1 \tan \theta \quad (185)$$

$$P_2 = S_1 \tan \theta \quad (186)$$

c. ICØND(J) = 3

For the condition ICØND(J) = 3, the current profile section is assumed to be circular and tangent to an adjoining straight section which is nearer the throat. The abscissa of the point of tangency is assumed to be PAR(2,J) (inches). As illustrated in figure 30, there are again four cases to be considered. As in the preceding case, the quantities S_1 and S_2 , equations (181) and (182), are used to treat all four cases in a single set of formulas.

In the present case, the coordinates of the point of tangency ("1" in figure 30) are known, since

$$X_1 = 2.54 \cdot \text{PAR}(2,J) \quad (187)$$

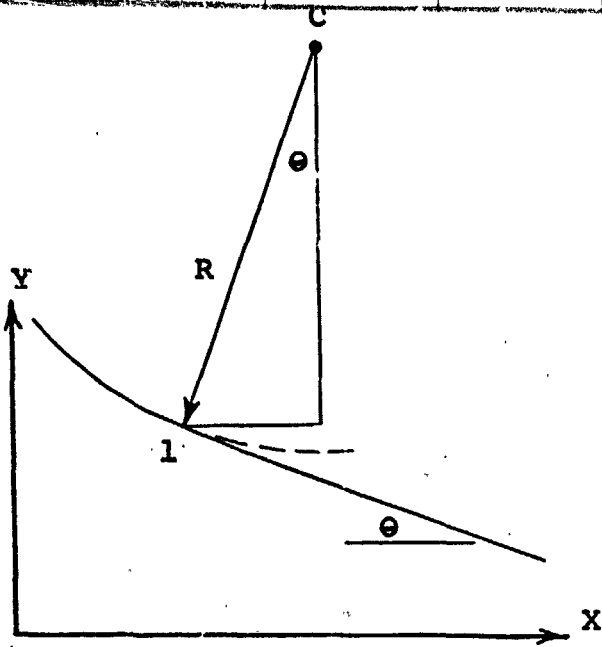
and Y_1 can be calculated from equation I(121) using the known parameter values for the adjoining straight section nearer the throat. The basic problem is to determine the coordinates (X_C , Y_C) of the circle center. Equations (183) are again applicable, and give

$$X_C = X_1 + S_1 S_2 R \sin \theta \quad (188a)$$

$$Y_C = Y_1 - S_2 R \cos \theta \quad (188b)$$

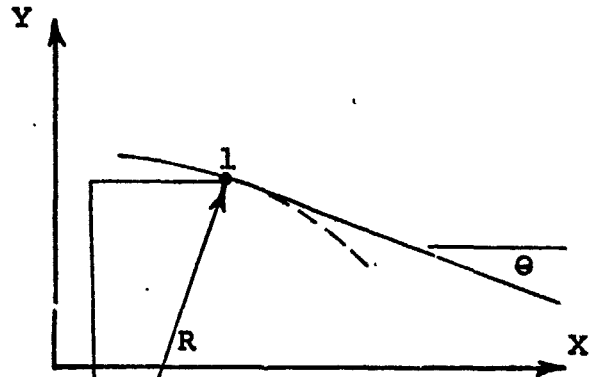
From equations I(122) and I(123), $P_2 = X_C$ and $P_1 = Y_C$. The radius P_3 is

①



(a)

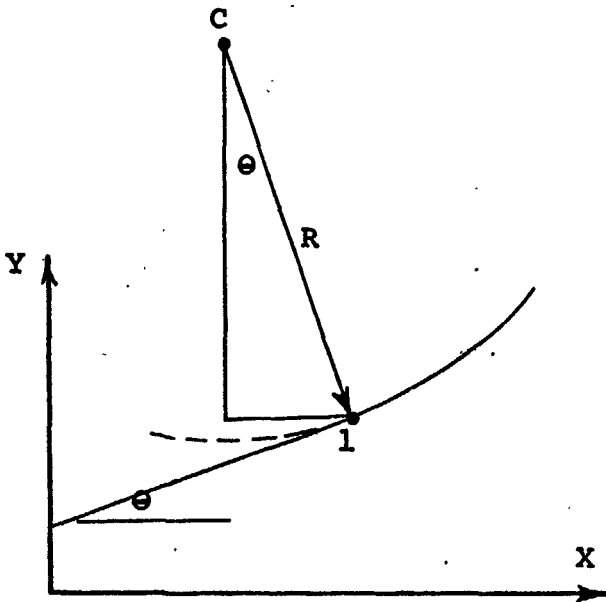
Upstream, ISHAPE(J) = 2



(b)

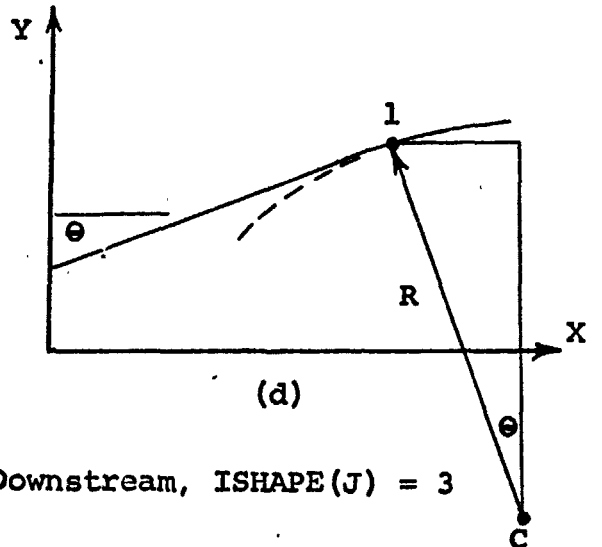
Upstream, ISHAPE(J) = 3

②



(c)

Downstream, ISHAPE(J) = 2



(d)

Downstream, ISHAPE(J) = 3

Figure 30. Geometry of the Condition ICØND(J) = 3

③

$$P_3 = 2.54 \cdot \text{PAR}(1, J) \quad (189)$$

d. $\text{ICOND}(J) = 4$

The condition $\text{ICOND}(J) = 4$ allows the determination of a circular section which is tangent to two adjacent straight sections (J+1 and J-1). Figure 31 illustrates the four cases which must be considered. The straight section nearer the throat has the index INDEX; the one farther from the throat has the index INDX2. The angles $\theta_1 = \text{PAR}(1, \text{INDEX})$ and $\theta_2 = \text{PAR}(1, \text{INDX2})$ are known and the abscissa $X_1 = 2.54 \cdot \text{PAR}(2, J)$ of the intersection of the two lines is given. Since the curvefit parameters for the section INDEX have already been determined, the ordinate Y_1 of the intersection can be calculated from eq. I(121).

Since the circular section to be determined is tangent to both lines, its center C lies on the bisector of the angle between the two lines. The length of this bisector, from the intersection point "1" to the circle center "C", is denoted by L (or XL in the Fortran). Also, the angle between the bisector and either of the lines is denoted by ϕ . This angle is given by

$$\phi = \frac{1}{2} (\pi - |\theta_2 - \theta_1|) \quad (190)$$

since $(\pi - |\theta_2 - \theta_1|)$ is the included angle between the lines. Then, from either of the right triangles with a vertex at C,

$$L = R / \sin \phi \quad (191)$$

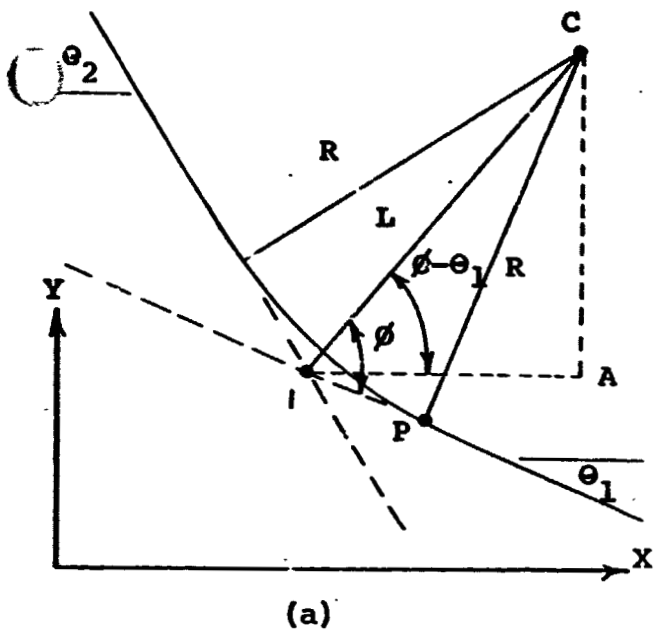
For $\text{ISHAPE}(J) = 2$ (parts (a) and (c) of figure 31), the angle CIA is equal to $\phi - \theta_1$. For $\text{ISHAPE}(J) = 3$ (parts (b) and (d) of figure 31), angle CIA is equal to $\phi - \theta_2$. Hence, for $\text{ISHAPE}(J) = 2$,

$$X_C = X_1 - S_1 L \cos (\phi - \theta_1) \quad (192a)$$

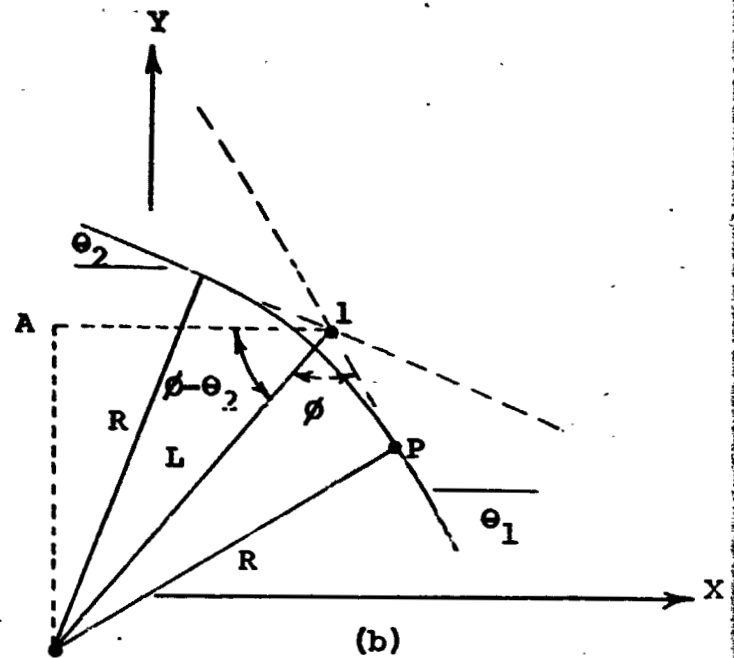
$$Y_C = Y_1 + L \sin (\phi - \theta_1) \quad (192b)$$

For $\text{ISHAPE}(J) = 3$,

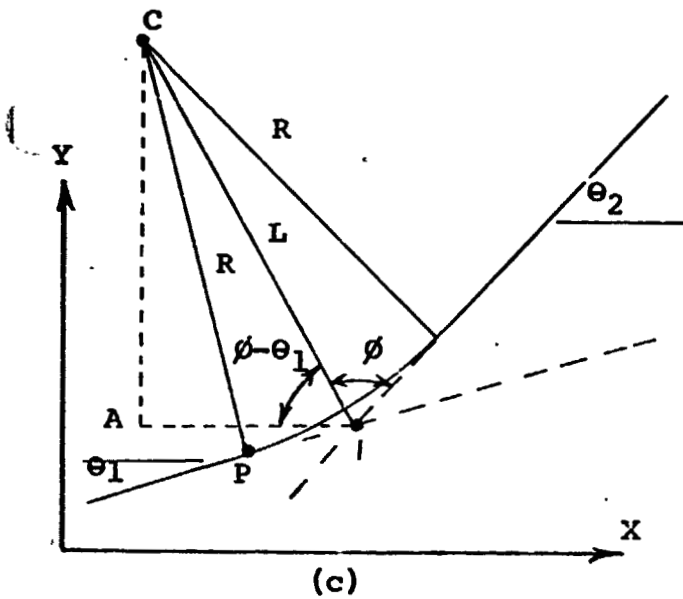
$$X_C = X_1 + S_1 L \cos (\phi - \theta_2) \quad (193a)$$



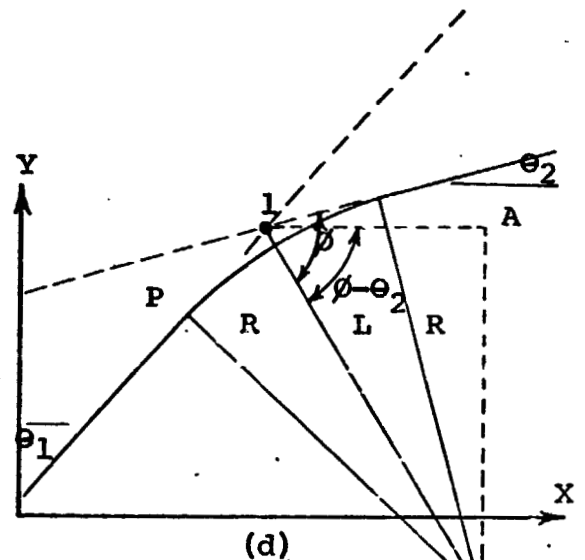
Upstream, ISHAPE(J) = 2



Upstream, ISHAPE(J) = 3



Downstream, ISHAPE(J) = 2



Downstream, ISHAPE(J) = 3

Figure 31. Geometry of the Condition. COND(J) = 4

$$Y_c = Y_1 - L \sin (\phi - \theta_2) \quad (193b)$$

For either case, the distance from point "1" to the junction point "P" between the circular section and the straight section nearer the throat is equal to $L \cos \phi$. The projection of this distance onto the X-axis is $L \cos \phi \cdot \cos \theta_1$. Hence, the abscissa of the junction point is

$$X_p = X_1 - S_1 L \cos \phi \cos \theta_1 \quad (194)$$

It is clear from figure 31 that ISHAPE(J) can be equal to 2 only if $\theta_2 > \theta_1$ and ISHAPE(J) can be equal to 3 only if $\theta_1 > \theta_2$. If the user inputs a value for ISHAPE(J) violating these conditions, an error message is written and execution is terminated.

(2) Input and Output. -- NATA and its auxiliary programs are designed for operation on either a UNIVAC 1108 or an IBM System 360. So far as NØZFIT is concerned, the principal difference between the two computer systems is the word length. The UNIVAC 1108 contains 36-bit words which can store 6 alphanumeric characters. The IBM 360 contains 32- and 64-bit words which can store 4 or 8 alphanumeric characters, respectively. The plot routines used at NASA/JSC on the UNIVAC 1108 and at Avco Systems Division on the IBM 360/75 assume that arrays of alphanumeric information are close-packed (i.e., with no unused characters). To provide correct handling of the information on the "header" card*, NØZFIT contains read and write statements based on both the A6 format (for the UNIVAC) and the A4 format (for the IBM 360). The correct input/output statements for alphanumeric information are selected by reference to an indicator IWØRDL, which is set in subroutine FRAMEB (see below). Two versions of FRAMEB are available, one for use on the UNIVAC 1108 at the Johnson Space Center, the other for use on the IBM 360/75 at Avco. Apart from the differences in this one subroutine, NØZFIT

*See discussion of NØZFIT inputs in Appendix D of Volume II (ref. 2).

and its plot subroutines are fully compatible with both computer systems.

The coding of the printed and punched output of the profile curvefit parameters is straightforward. In the profile calculation (statements 240 to 310), the abscissa X is varied by increments of 0.03 inch from X = XSTART to X = XSTART + 6 inches. At each X, the index K of the profile section in which X lies is determined by finding the first K for which the downstream boundary ATP(K) is greater than X. The ordinate Y is then computed from equation I(121), I(122), or I(123) according as ISHAPE(K) is equal to 1, 2, or 3. The result of this calculation is in centimeters, and is converted into inches for output.

Finally, if the indicator PLOTS is .TRUE., the profile is plotted by a call to subroutine GRAPH. After the final case in the job (ENDJOB = .TRUE.), the entry FRAMB2 of subroutine FRAMEB is called. If the program is being run on the Avco IBM 360/75, FRAMB2 calls a library routine which closes the plot data set.

A.1.2 Subroutine FRAMEB

Subroutine FRAMEB contains all of the irreconcilable differences between the UNIVAC 1108 and IBM 360 versions of NØZFIT. Two versions of this routine are provided (Section A.3), one for each type of computer system.

The IBM 360 version sets the indicator IWØRDL to 2, to induce NØZFIT to use A4 formats in reading and writing alphanumeric information. It also calls the Avco library routine IDFRMV to produce an identification frame preceding the plots of nozzle profiles, if the control variable PLOTS is .TRUE.. When called through the entry FRAMB2, it calls the Avco library routine PLTND to empty the plot buffer and close the plotting data set.

The UNIVAC 1108 version of FRAMEB sets IWØRDL to 1, to induce NØZFIT to use A6 formats in reading and writing alphanumeric information. It performs no other function.

--	--	--	--	--	--	--	--

A plot identification frame is produced automatically on the 1108 computers at NASA/JSC when plots are requested on the JØB card, and the data set is also closed automatically.

A.1.3 Subroutine GRAPH

GRAPH is a general-purpose plotting routine which was developed for use in the auxiliary programs to the NATA code. It uses several lower level plot routines programmed by North American Aviation (NAA) as software for the S-C 4020 high-speed microfilm recorder. The capabilities of GRAPH may be summarized as follows:

- (1) It produces Cartesian graphs of Y versus X from data stored in arrays.
- (2) The data points can be displayed by up to 10 different plot symbols, or not displayed.
- (3) The data points can be connected by lines, or not connected. The lines can be continuous, dashed (with user-controlled dash length), dotted, or traced by a string of Charactron symbols.
- (4) Different data sets can be plotted on the same frame. The points from the different data sets can be represented by different plot symbols and/or connected by lines of different types.
- (5) A parameter value can be associated with each data set. The curves for different data sets can be labelled with numerical labels and a table displaying the labels and the associated parameter values can be shown in the margin to the right of the graph. Up to three distinct parameters, each with up to 10 values, can be included in a single plot.

The arguments in the calling sequence of GRAPH are defined in the Glossary of Fortran Symbols (Section A.2).

If a table of parameter values is to be printed in the frame, the first call to GRAPH must be preceded by a call to the NAA routine SETMIV (references 3, 17) to provide the required space in the right margin:

(ALL SETMIV (30, 125, 24, 24)

(1) Grid generation. -- The first part of GRAPH, down to statement 110, is executed only when the routine is called with $NF \neq 0$ to establish the grid for a new frame. In this case, the counter JCURVE for data sets plotted and the counter NPL for parameter tables are initialized to zero, and the NAA routine SMXYV (references 3, 17) is called to establish the logarithmic or linear mode for each axis as indicated by the GRAPH arguments LX, LY. Then the NAA routine DXDYV (references 3, 17) is called for each axis to determine the arguments DX, DY, NNM, MMM, III, JJJ, NXX, NYY of subroutine GRIDIV. If either axis is to be used as a logarithmic scale ($LX \neq 0$ or $LY \neq 0$), then the corresponding call to DXDYV is skipped, and DX and DY are set to 1 in accordance with instructions in references 3 and 17.

Beginning at statement 50, the parameters NXY(1) and NXY(2), which determine the numbers of characters to be displayed in the labels of vertical and horizontal grid lines, are established. The corresponding values NXX and NYY provided by DXDYV were found to be unsatisfactory in some applications. For each axis, if \log_{10} of the maximum absolute value to be plotted is greater than 5 or less than -3, the grid-line labels are printed in scientific notation, with three significant digits, by setting $NXY(I) = -3$. Otherwise, a fixed-point format is used, with a number of characters determined from the magnitude of the maximum value to be plotted.

Next, the GRIDIV arguments III and JJJ are reversed in sign to force placement of the grid line labels outside

of the grid area. Also, if numerical labels for the individual curves are to be displayed, the lower limits XL and YL are decreased slightly, and the upper limit YU is increased slightly, to allow space for the labels within the grid area. Then GRIDIV is called to draw the grid and to establish scaling factors for subsequent use in plotting. Finally, the counter ICURVE for curves in the first family (i.e., for various values of the first parameter, if any) is initialized to zero.

(2) Plotting of a data set. -- The second part of GRAPH plots an individual data set, that is, the data contained in the arrays X and Y for a single call to GRAPH. This section of the subroutine extends from statement 110 down to statement 190.

First, the counters ICURVE and JCURVE are each incremented by 1. If ICURVE is equal to 1 after this increase, the counter NPL for parameter tables is also increased by 1. Note that ICURVE is equal to 1 at this point if this is the first call to GRAPH for a new frame, or if ICURVE was set to zero in the calling routine before the current call to GRAPH. Setting ICURVE to 0 in the calling routine is a signal to GRAPH that a new family of curves with a new parameter is being started.

Next, the NAA routines NXV and NYV are used to initialize IX1 and IY1 to the raster coordinates corresponding to X(1) and Y(1), respectively. Then the plot symbol to be used in displaying the data points is selected on the basis of the arguments NC(1) and NC(2). If NC(1) is negative, the symbol index NSYMBL is set equal to ICURVE, so that a new symbol from the standard table in PØINTB is used for the points on each successive curve in a family of curves. If NC(1) is positive, NSYMBL is set equal to NC(1); this option allows direct user control of the plot symbol. For NC(1) = 0, the individual points are not plotted.

The $\text{D}\emptyset$ loop from statement 140 to statement 180 plots the data points and draws the connecting lines, if any. First, subroutine $\text{P}\emptyset\text{INTB}$ (Section A.1.4 of this appendix) is called to plot the current (Ith) data point. Then I2 is set equal to I+1. If $\text{LINE}(1)$ is nonzero, subroutine LINEB (Section A.1.5 of this appendix) is called to draw the line segment from the Ith to the I2th point. The type of line to be drawn (continuous, dashed, etc.) is determined by the argument LINE in the call to LINEB . Then IX1 and IX2 are set equal to the raster coordinates corresponding to $\text{X}(I2)$ and $\text{Y}(I2)$, respectively, using the NAA function routines NXV and NYV , for use in calling $\text{P}\emptyset\text{INTB}$ the next time around the loop.

When these plotting operations have been completed, if JCURVE is equal to 1, the NAA routine CHSIZV is called with arguments (3,3) to restore the normal character size.* Then subroutine RITE2V (references 3, 17) is called to write the abscissa and ordinate labels ABSL and ORDL and the plot title TITL in the bottom, left, and top margins of the plot, respectively.

(3) Parameter tables and labels. -- If the argument LL is zero, then parameter tables and curve labelling are not to be used, and a RETURN is executed at statement 190 in GRAPH . If LL is 1, the parameter tables are produced but the curve labelling is skipped. If LL is neither 0 nor 1, then the curves are labelled also. To implement the labelling, the code calls the NAA routines NXV and NYV to determine the raster coordinates, NXV1 and NYV1 , of the first point on the curve. It then computes the raster coordinates, IX1 and IY1 , of a point which lies 16 raster units away from the first point of the curve in a direction opposite to that in which the second point lies. It then writes the curve label ($\text{CURVE} = \text{ICURVE}$) into the plot by calling the NAA subroutine LABLV (references 3, 17).

*A nonstandard character size is set in $\text{P}\emptyset\text{INTB}$ by calling $\text{CHSIZV}(3,2)$, to obtain plot symbols whose height and width are approximately equal.

The tables of parameter values are produced by the coding from statement 200 to statement 300. For ICURVE = 1, the code first writes the table heading. The heading consists of a sample of the type of line used for the current family of curves, and a two-line printed specification of the nature of the current parameter. The sample line is produced by calling the entry LINEBR of subroutine LINEB (discussed below). It is horizontal and lies at the raster ordinate IY2. The printed heading is produced by calling the NAA routine PRINTV. Its first line lies 15 raster units below the sample line, and its second line is an additional 20 raster units lower. The tables for different parameters (families of curves) have a vertical separation of 300 raster units.

The body of the parameter table consists of rows each containing the numerical label for the curve (equal to CURVE), a sample of the plot symbol used, and the parameter value. The numerical label is omitted for LL = 1. When it is included, it is produced by calling LABELV. The sample plot symbol is produced by calling POINTB, and the parameter value by another call to LABELV. The arguments NCHAR and NDMAX of LABELV are determined as functions of IPVM, which is $\log_{10}(\text{PVAL})$ rounded down in magnitude to the nearest integer. If IPVM is greater than 4 or less than -2, NCHAR is set to -3 to give scientific notation with three significant figures. Otherwise, fixed-point notation is used, and NCHAR and NDMAX are adjusted to provide the required number of characters and positions to the left of the decimal point.

A.1.4 Subroutine LINEB

Subroutine LINEB draws a straight line between two specified points (X1, Y1), (X2, Y2). When it is called through its entry LINEBR, the two points are specified in raster coordinates (MX1, MY1), (MX2, MY2). The argument LLINE is a three-element array whose values control the type of line produced and its darkness, as defined in the Glossary of Fortran Symbols (Section A.2).

① The raster coordinates of the two points to be connected by a line are denoted, within LINEB, by (NX1, NY1), (NX2, NY2). If entry was through LINEBR, these values are set equal to MX1, etc. If entry was through LINEB, the raster coordinates are determined by calling the NAA subroutines NXV and NYV.

For LINE \equiv LLINE(1) = 1, the points are connected with a continuous line by calling NAA subroutine LINEV. This call is executed IDARK times to obtain the desired darkness.

① For LINE > 1, beginning in statement 50, the sine (SINA) and cosine (COSA) of the angle between the vector from point 1 to point 2 and the positive X-axis are computed. Then, for LINE = 2, a dashed line is drawn from point 1 to point 2, with the dash length in millimeters* being approximately equal to IPAR \equiv LLINE(3). The distance DEL from the beginning of one dash to the beginning of the next is calculated, in raster units, as $10.8 \times \text{IPAR}$. The numerical coefficient in this formula is obtained by noting that the frame width is 1024 raster units and is 190 mm in hard copy. Thus, 1 mm on the hard copy is equivalent to 5.4 raster units, the dash length HDEL is equal to $5.4 \times \text{IPAR}$, and DEL is twice as large, allowing for the gap between two successive dashes. The number of dashes to be used in the line joining the two points is thus $\text{NP} = \text{DIST}/\text{DEL}$, where DIST is the separation of the points in raster units. If the two points are too close together in comparison with the specified dash length, integer arithmetic gives $\text{NP} = 0$. This possibility should be borne in mind by the user of GRAPH.

LINEB next computes the increments IDX and IDY in raster coordinates between the beginning and end of a single dash, using the previously determined trigonometric

*This description of the dash length is applicable to hard copy from the Calcomp 890 CRT Plotter at Avco, or to enlargements of microfilm plots produced on the SD 4060 at NASA/JSC to a frame size of 7.5 x 7.5 inches.

functions CØSA and SINA. It then calls the NAA subroutine LINE2V (references 3, 17) NP times to draw the dashed line joining the end points specified in the arguments. In the DØ loop, DØ 70 I = 1, NP, the quantities VX1 and VY1 represent the desired starting points for each dash in raster coordinates. These quantities are calculated using floating point arithmetic to avoid the loss of accuracy possible in integer arithmetic, and are rounded down to obtain the integer values NX1, NY1 used in the calls to LINE2V.

For LINE = 3, the line joining the specified end points is produced by plotting a string of Charactron symbols. The symbol to be used is specified by IPAR, which is used as the index in the POINTV character table (references 3, 17). The following list is excerpted from references 3 and 17.

<u>IPAR</u>	<u>Character</u>
0	.
1	O
2	x
4	Y
5	+
6	*
10	H
15	⊗
22	ø
25	I

The separation of successive points, DEL, is set to 5 raster units if IPAR = 0 (i.e., if the plotting character is a dot), and is set to 15 raster units otherwise. The number of characters to be used in tracing the line is then NP = DIST/DEL. The string of characters is then produced by calling the four-argument version of PØINTV (references 3, 17) NP times, in the loop DØ120 I = 1, NP. The minus sign on the argument IPAR of PØINTV suppresses the plotting of a central dot in each character.

A.1.5 Subroutine POINTB

This routine plots a point at the raster coordinates (IX, IY), using a plot symbol selected from a table ICHAR(I) of ten standard symbols. The index NSYMBL in the calling sequence specifies which symbol is to be used. The symbols are as follows:

<u>I</u>	<u>Symbol</u>
1	X
2	+
3	O
4	*
5	H
6	C
7	∅
8	Y
9	λ
10	I

These symbols are produced by calling the NAA subroutine RITE2V (references 3, 17). The inversions for I = 7 and 9, and the 90-degree rotation for I = 10, are accomplished using the argument K(NSYMBL) of RITE2V. Before RITE2V is called, a call is made to CHSIZV (references 3, 17) to make the height of the symbols nearly equal to their width instead of 50 percent greater, as is normally the case. The argument IDARK of POINTB and RITE2V controls the number of times the symbol is overdrawn and thus its darkness in the plot.

A.2 Glossary of Fortran Symbols for NOZFIT

This section defines the Fortran symbols used in the NOZFIT main program and in its associated subroutines. The variables are listed in alphanumeric order within each routine, and the routines themselves are in alphabetic order. There are no common variables.

A.2.1 Main Program

ABSL(I) Array containing alphanumeric abscissa label for plots

ACØN Conversion factor ($180/\pi$) from radians to degrees

ATP(JP) Axial coordinate at the downstream boundary of the JPth profile section (cm)

CARDS Logical control; .FALSE. value suppresses punched card output

CØMMNT(I) Hollerith descriptive data

CTHETA Cos (THETA)

DTH Throat diameter (inch)

DX Increment in X for calculation of profile (inch)

ENDJØB Logical control; .TRUE. value terminates the job at the end of the current case

FISHAP(I) Floating point value representing ISHAPE(I) in punched-card output

FNSECT(I) NSECTS(I) + 0.1 for card output

I DØ index

IC ICØND(J) for the Jth profile section

ICØND(J) Index specifying the type of condition to be used in determining the Jth profile section (see input discussion, Appendix D of Volume II, ref. 2 or Section A.1.1 of the present Appendix)

ICURVE Unused argument of subroutine GRAPH

INCR(L) 0 for L = 1; 1 for L = 2

INDEX	Index of the profile section adjacent to the current (Jth) section and nearer the throat
INDX2	Index of the profile section adjacent to the current (Jth) section and farther from the throat
ISGN(L)	1 for L = 1; -1 for L = 2
ISH	ISHAPE(INDEX)
ISHAPE(J)	Indicator for shape of Jth profile section (see input discussion, Appendix D of Volume II)
ISJ	ISHAPE(K)
IWØRDL	Indicator for number of characters per word on the type of computer system being used: IWØRDL = 1 6-character word (UNIVAC 1108) IWØRDL = 2 4-character word (IBM 360)
J	Profile section index
JP	Index of profile section boundaries
J1	2K - 1; first J value on the Kth card containing PARAM(I,J) data in the punched output
J2	J1 + 1; second J value on the Kth card containing PARAM(I,J) data in the punched output
K	DØ index; profile section index in calculation of profile from curvefit
KP	Continuation card counter punched in column 6 of output cards

L	Indicator for upstream profile sections (L = 1) and downstream sections (L = 2)
LINE(I)	Argument of subroutine GRAPH (see below)
LL	Argument of subroutine GRAPH (see below)
NC(I)	Argument of subroutine GRAPH (see below)
NØZZLE	Profile index for use in NATA
NS	NSECTS(L) in computation of profile fit parameters; total number of sections in coding of output
NSM1	NS-1
NSECTS(L)	Number of profile sections upstream (L = 1) and downstream (L = 2) of the throat
ØRDL(I)	Alphanumeric label for ordinate in plot
PAR(I,J)	Input parameter values for the Jth profile section; see discussion of inputs, Appendix D of Volume II
PARAM(I,J)	Parameters of curvefit for Jth profile section; see discussion of outputs, Appendix D of Volume II
PHI	$0.5 (\pi - \text{THETA2} - \text{THETA1})$; see discussion of ICØND = 4 in Section A.1.1.
PI	3.14159265
PLABL(I)	Argument of subroutine GRAPH (see below)
PLØTS	Logical control, .FALSE. value suppresses plot output
PVAL	Argument of subroutine GRAPH (see below)

R	Circle radius for a circular-arc section (cm)
RTHCM	Throat radius in centimeters
STHETA	Sin (THETA)
S1	-1.0 for L = 1, +1.0 for L = 2
S2	0.0 for ISHAPE(J) = 1; -1.0 for ISHAPE(J) = 2; +1.0 for ISHAPE(J) = 3
S2ARR(ISH)	Array containing S2 values for the three shapes
THETA	Angle of inclination of a straight section to the nozzle axis (radians)
THETA1	For ICØND = 4, inclination angle of the adjacent section closer to the throat
THETA2	For ICØND = 4, inclination angle of the adjacent section farther from the throat
TITLE(I)	Title for plots (Hollerith)
TTHETA	Tan(THETA)
X	Axial coordinate (cm)
XC	X-coordinate of the circle center for the adjacent section nearer the throat
XINCH(I)	Axial coordinate (inches)
XL	$R/\sin(\text{PHI})$; see discussion of ICØND = 4 in Section A.1.1
XLØ	Lower limit of abscissa in plot
XSTART	Left limit on X for calculating nozzle profile (inch)
XUP	Upper limit of abscissa in plot

XZERØ	Axial coordinate of start of boundary layer, for use in NATA (cm)
XZERØI	XZERØ expressed in inches
XI	Intermediate X values in calculations of parameters for ICØND = 2, 3, and 4
Y	Profile ordinate (cm)
YC	Y-coordinate of the circle center for the adjacent section nearer the throat
YINCH(I)	Profile ordinate (inches)
YLØ	Lower limit on ordinate in plot
YUP	Upper limit on ordinate in plot
YI	Intermediate Y values in calculations of parameters for ICØND = 2, 3, and 4

A.2.2 Subroutine FRAMEB

BIN	Alphanumeric bin number for plot identification frame (Avco)
IWØRDL	Defined above (Main program)
PLØTS	Defined above (Main program)
XMEMØ	Alphanumeric memo number for plot identification (Avco)
XNAME	Alphanumeric submitter name for plot identification frame (Avco)

A.2.3 Subroutine GRAPH

ABSL Array containing 48-character label for abscissa scale

CURVE ICURVE expressed as a floating-point value

DC Parameter controlling the spacing of grid lines on linear scales; DC = 16 is a normal value. The spacing increases or decreases with the value of DC

DX Floating point data increment at which vertical grid lines are displayed

DY Floating point data increment at which horizontal grid lines are displayed

DYL Amount by which YL is decreased and YU increased to allow space for numerical labels of curves

HYP Distance (in raster units) between the first two points of a curve

I \emptyset index

ICURVE Counter for the number of curves plotted on the same frame. If this argument is set to zero by the calling program, a new parameter label for a new series of curves is printed in the margin; this feature is used only when two or more families of curves are to be plotted in the same frame.

IDXV Interval in raster units between the abscissas of the first two points in a curve

IDYV Interval in raster units between the ordinates of the first two points of a curve

IERR	Error indicator in subroutine DXDYV (references 3, 17); if a nonzero value of IERR is returned, grid generation is impossible with the given upper and lower limits on X and Y
III	Argument of subroutine GRIDIV (references 3, 17). Every IIIth vertical line of the grid is labelled with a numerical value; a negative III forces placing the labels outside the grid
IMAG(I)	For I = 1 and 2, \log_{10} of the largest numerical value to be plotted on the abscissa and ordinate, respectively
IPVM	\log_{10} of the absolute magnitude of a parameter value (PVAL)
IXI	Abscissa of the location where a curve label is to be printed (raster units)
IY	Ordinate of the location where a parameter value is to be printed in the margin (raster units)
IY1	Ordinate of the location where a curve label is to be printed (raster units)
IY2	Ordinate of the location where a sample of the type of line used for a family of curves is drawn in the margin (raster units)
I2	$i + 1$
J	D ϕ index
JCURVE	Counter for the total number of curves plotted in a given frame

--	--	--	--	--	--

JJJ Argument of subroutine GRIDLV (references 3, 17). Every JJJth horizontal line of the grid is labelled with a numerical value; a negative JJJ forces placing the labels outside the grid

LINE(1) Control parameters for lines joining data points, defined as follows:

LINE(1)

- 0 Do not join points with lines
- 1 Join points with continuous lines
- 2 Join points with dashed lines
- 3 Join points with strings of characters

LINE(2) Number of strikes for line; use 2 for dark lines

LINE(3) Ignored if LINE(1) is 0 or 1
Length of dashes in mm if LINE(1) = 2
Index of character in POINTV table if LINE(1) = 3 (see Procedures Manual, Computation and Analysis Division, MSC, p. 5.c.7.5.1.127)
(Use LINE(3) = 0 with LINE(1) = 3 to produce dotted lines)

LL Control parameter for labelling of curves:

- LL = 0 Suppress labelling of curves and printing of parameter values in table at right of grid
- LL = 1 Print parameter values and plot symbols but do not label curves
- LL = 2 Label curves and print parameter values and plot symbols

LX Argument of subroutine SMXYV (references 3, 17); 0 sets linear mode for abscissa, nonzero log mode

O	<p>LY Argument of subroutine SMXYV (references 3, 17); 0 sets linear mode for ordinate, nonzero log mode</p>
1	<p>MMM Argument of subroutine GRIDLV (references 3, 17). Every MMMth horizontal grid line is retraced for emphasis</p>
	<p>NC(I) For I = 1 and 2, control parameters for plotting points:</p> <p style="padding-left: 40px;">NC(1) < 0 Use standard table of 10 characters for plotting points, a new character for each successive call to GRAPH for the same family of curves in the same frame (see subroutine PØINTB)</p> <p style="padding-left: 40px;">NC(1) = 0 Suppress plotting of points</p> <p style="padding-left: 40px;">NC(1) > 0 Use the NC(1)th character in the standard table (see PØINTB)</p> <p style="padding-left: 40px;">NC(2) Number of strikes for each symbol plotted; use 1 for light characters, 2 for dark</p>
	<p>NCHAR Argument of subroutine LABLV (see references 3, 17); number of characters to be displayed, including leading blanks and the decimal point, if any</p>
	<p>NC1 NC(1)</p>
	<p>NC2 NC(2)</p>
	<p>NDMAX Argument of subroutine LABLV (see references 3, 17); maximum number of characters to be displayed to the left of the decimal point</p>

NF	Control parameter for frame advance: NF = 0 Plot data on same frame NF = 1 Start a new frame, establish a grid, and plot the data on the new frame
NLAST	Argument of subroutine RITE2V (references 3, 17); not used in GRAPH
NNN	Argument of subroutine GRIDIV (references 3, 17). Every NNNth vertical grid line is retraced for emphasis
NP	Number of data points to be plotted
NPL	Counter for families of curves plotted in the same frame
NSYMBL	Index of point symbol in standard table (see POINTB)
NXV1	Abscissa of first point in raster coordi- nates
NXX	Argument of subroutine DXDYV (references 3, 17); not used in GRAPH
NXY(I)	For I = 1 and 2, the number of characters to be displayed in the labels of vertical and horizontal grid lines; arguments of subroutine GRIDIV (references 3, 17)
NYV1	Ordinate of first point in raster coordi- nates
NYY	Argument of subroutine DXDYV (references 3, 17); not used in GRAPH
ØRDL(I)	Alphanumeric label for ordinate
PLABL(I,J)	For J = 1, 2, a two-line alphanumeric label for the parameter

PVAL	Parameter value
TITL(I)	Alphanumeric title
X(I)	Array containing the abscissas of the data points
XL	Lower limit on X for all data to be plotted in a given frame
XU	Upper limit on X for all data to be plotted in a given frame
XYMAG(I)	For I = 1 or 2, the largest absolute magnitude abscissa or ordinate value to be plotted in a given frame
Y(I)	Array containing the ordinates of the data points
YL	Lower limit on Y for all data to be plotted in a given frame
YU	Upper limit on Y for all data to be plotted in a given frame
A.2.4 Subroutine LINEB	
CØSA	Cosine of the angle between the vector to be drawn and the positive X-axis
DEL	Twice the dash length in raster units
DIST	Distance between the beginning and end of the vector in raster units
HDEL	Dash length in raster units
I	DØ index over the dashed or plot symbols used to draw the vector
IDARK	Number of times lines or symbols are drawn

O	IDX	Increment in the abscissa for a dash (raster units)
	IDY	Increment in the ordinate for a dash (raster units)
	IPAR	Length of dashes in millimeters for LINE = 2; integer (NS) for selecting the plot symbol from the table in subroutine PØINTV (references 3, 17) for LINE = 3
	J	DØ index for double-strike plotting to obtain dark lines or symbols
	LINE	Control parameter for type of line: LINE = 1 Continuous line LINE = 2 Dashed line LINE = 3 String of Charactron symbols
	LLINE(I)	LLINE(1) = LINE LLINE(2) = IDARK LLINE(3) = IPAR
	MX1	Abscissa of the start point in raster units
	MX2	Abscissa of the end point in raster units
	MY1	Ordinate of the start point in raster units
	MY2	Ordinate of the end point in raster units
	NP	Number of dashes in vector
	NX1	Abscissa of the start point in raster units
	NX2	Abscissa of the end point in raster units
	NY1	Ordinate of the start point in raster units
	NY2	Ordinate of the end point in raster units

RASTER	Logical indicator, .TRUE. for entry through LINEBR, .FALSE. for entry through LINEB
SINA	Sine of the angle between the vector to be drawn and the positive X-axis
SQ	Square of the distance between the start and end points of the vector (raster units)
VX1	Floating point value of NX1
VY1	Floating point value of NY1
X1	Abscissa of the start point
X2	Abscissa of the end point
Y1	Ordinate of the start point
Y2	Ordinate of the end point

A.2.5 Subroutine PØINTB

CHAR(I)	Array of characters for plotting points
IDARK	Argument of RITE2V controlling darkness of plotted symbol (see references 3, 17)
IX	Abscissa of location where point is plotted (raster coordinate)
IY	Ordinate of location where point is plotted (raster coordinate)
K(I)	Argument of RITE2V controlling orientation of plotted symbol; used to present each of the symbols "H", "C", and "Y" in two orientations
NLAST	Argument of RITE2V (references 3, 17); not used in PØINTB

NSYMBL

Index used to select plotting symbol from
CHAR array

A.3 Listing of NØZFIT

Presented below are source program listings for the main program of NØZFIT, and for subroutines FRAMEB, GRAPH, LINEB, and PØINTB. Listings are provided for both the IBM (Avcc) and UNIVAC (NASA/JSC) versions of FRAMEB.


```

ATP(JP)=X1
GO TO 130
IF (ISHAPE(J).NE.2.AND.ISHAPE(J).NE.3) GO TO 120
IF (ISHAPE(INDEX).NE.1) GO TO 120
CONDITION ICOND=3
R=2.54*PAR(1,J)
X1=2.54*PAR(2,J)
IC=ISHAPE(J)
S2=S2ARR(ISH)
THETA=PAR(1,INDEX)/ACON
Y1=PARAM(1,INDEX)+PARAM(2,INDEX)*X1
PARAM(1,J)=Y1-S2*R*COS(THETA)
PARAM(2,J)=X1+S1*S2*R*SIN(THETA)
PARAM(3,J)=R
ATP(JP)=X1
GO TO 130
IF (ISHAPE(J).NE.2.AND.ISHAPE(J).NE.3) GO TO 120
IF (ISHAPE(INDEX).NE.1) GO TO 120
INDEX=J-1SGN
IF (ISHAPE(INDX2).NE.1) GO TO 120
CONDITION ICOND=4
R=2.54*PAR(1,J)
X1=2.54*PAR(2,J)
THETA1=PAR(1,INDEX)/ACON
THETA2=PAR(1,INDX2)/ACON
PHI=0.5*(PI-ABS(THETA2-THETA1))
IF (THETA2.GT.THETA1.AND.ISHAPE(J).EQ.3) GO TO 120
IF (THETA2.LT.THETA1.AND.ISHAPE(J).EQ.2) GO TO 120
Y1=PARAM(1,INDEX)+PARAM(2,INDEX)*X1
XL=R/SIN(PHI)
ATP(JP)=X1-S1*XL*COS(PHI)*COS(THETA1)
PARAM(3,J)=R
IF (ISHAPE(J).EQ.3) GO TO 110
PARAM(2,J)=X1-S1*XL*COS(PHI-THETA1)
PARAM(1,J)=Y1+XL*SIN(PHI-THETA1)
GO TO 130
PARAM(2,J)=X1+S1*XL*COS(PHI-THETA2)
PARAM(1,J)=Y1-XL*SIN(PHI-THETA2)
GO TO 130
WRITE (6,350) J
CALL EXIT
CONTINUE
CONTINUE
OUTPUT
PRINTED OUTPUT
IF (I*ORDL.EQ.1) GO TO 150
WRITE (6,360) (COMMENT(I),I=1,18)
GO TO 160
WRITE (6,370) (COMMENT(I),I=1,12)
WRITE (6,380)
WRITE (6,390)
NS=NSECTS(1)+NSECTS(2)
DD 18C J=1,NS

```

NOZ 111
NOZ 112
NOZ 113
NOZ 114
NOZ 115
NOZ 116
NOZ 117
NOZ 118
NOZ 119
NOZ 120
NOZ 121
NOZ 122
NOZ 123
NOZ 124
NOZ 125
NOZ 126
NOZ 127
NOZ 128
NOZ 129
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NOZ 164
NOZ 165

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170 IF (J.EQ.NS) GO TO 170
180 WRITE (6,400) J,ATP(J),(PARAM(I,J),I=1,3)
C CONTINUE
IF (.NOT.CARDS) GO TO 240
PUNCHED OUTPUT AND PRINTED CARD IMAGES
RTHCM=1.27*DTH
XZERO=2.54*XZERIC
DO 190 I=1,2
FNSECT(I)=FNSECTS(I)+0.1
DO 200 I=1,12
IF (I.GT.NS) ISHAPE(I)=0
FISHAP(I)=ISHAPE(I)+0.1
WRITE (6,420)
IF (I.ORDL.EQ.1) GO TO 210
PUNCH 520, (COMMENT(I),I=1,16),NOZZLE
WRITE (6,440) (COMMENT(I),I=1,16),NOZZLE
GO TO 220
PUNCH 510, (COMMENT(I),I=1,11),NOZZLE
WRITE (6,430) (COMMENT(I),I=1,11),NOZZLE
PUNCH 530, NOZZLE,RTHCM,XZERO,(FNSECT(I),I=1,2),NOZZLE
WRITE (6,450) NOZZLE,RTHCM,XZERO,(FNSECT(I),I=1,2),NOZZLE
PUNCH 540, (FISHAP(I),I=1,12),NOZZLE
WRITE (6,460) (FISHAP(I),I=1,12),NOZZLE
PUNCH 550, (ATP(I),I=1,6),NOZZLE
WRITE (6,470) (ATP(I),I=1,6),NOZZLE
PUNCH 560, (ATP(I),I=7,11),NOZZLE
WRITE (6,480) (ATP(I),I=7,11),NOZZLE
DO 230 K=1,6
J1=2*K-1
J2=J1+1
KP=K+3
WRITE (6,490) KP,(PARAM(I,J),I=1,3),J=J1,J2),NOZZLF
PUNCH 570, KP,(PARAM(I,J),I=1,3),J=J1,J2),NOZZLE
PUNCH 580, COMMENT(1),NOZZLE
WRITE OUT OF PROFILE
WRITE (6,500) COMMENT(1),NOZZLE
WRITE (6,590)
WRITE (6,600)
DX=C.C?
DO 310 I=1,201
XINCH(I)=XSTART+(I-1)*DX
X=2.54*XINCH(I)
NSMI=NS-1
DO 250 J=1,NSMI
KEJ (ATP(J),GT.X) GO TO 260
CONTINUE
X=NS
ISJ=ISHAPE(K)
GO TO (270,280,290), ISJ
Y=PARAM(1,K)+PARAM(2,K)*X
GO TO 300
Y=PARAM(1,K)-SORT(PARAM(3,K)**2-(X-PARAM(2,K))**2)
250
260
270
280
290
300

```

```

290 GO TO 300
300 Y=PARAM(1,K)+SQRT(PAPAM(3,K)**2-(X-PARA(2,K))**2)
WRITE (6,610) XINCH(1),YINCH(1)
310 CONTINUE
IF (.NOT.PLOTS) GO TO 320
PLOT OF PROFILE
NC(1)=0
NC(2)=1
LINE(1)=1
LINE(2)=2
LL=0
XLO=XINCH(1)
XUP=XINCH(201)
YLO=0.
YUP=0.
CALL GRAPH (1,0,0,201,XINCH,YINCH,LO,XUP,YLO,YUP,16,NC,LINE,ORDL
320 1,AJSL,TITLE,LL,VAL,PLABL,ICURVE)
IF (.NOT.ENDJOB) GO TO 10
IF (PLOTS) CALL FRAMB2
CALL EXIT
330 FORMAT (18A4)
340 FORMAT (12A6)
350 FORMAT (20HC INCOMPATIBLE DATA FOR SECTION,I3)
360 FORMAT (1H,18A4//)
370 FORMAT (1H,12A6//)
380 FORMAT (2PH NOZZLE PROFILE PARAMETERS)
390 FORMAT (1HC,8X,1HJ,17X,3HATP,10X,10HPARAM(1,J),10X,10HPARAM(2,J),1
400 1X,10HPARAM(3,J))
410 FORMAT (110,1P4E20,6)
420 FORMAT (///20H DATA CARDS PRODUCED/)
430 FORMAT (1X,1HC,5X,11A6,1X,4HNOZZ,I3)
440 FORMAT (1X,1HC,5X,16A4,3X,4HNOZZ,I3)
450 FORMAT (7X,7HDATA ZP,12,2H /,F10,5,1H,,F10,5,1H,,F6,1,1H,,F6,1,1H,
1,20X,4HNOZZ,I3)
460 FORMAT (6X,1H1,6X,12(F4,1,1H,),1X,4HNOZZ,I3)
470 FORMAT (5X,1H2,6(F10,6,1H,),1X,4HNOZZ,I3)
480 FORMAT (6X,1H3,5(F10,6,1H,),12X,4HNOZZ,I3)
490 FORMAT (6X,1H4,6(F10,6,1H,),1X,4HNOZZ,I3)
500 FORMAT (6X,1H5,6(F10,6,1H,),1X,4HNOZZ,I3)
510 FORMAT (6X,1H6,6(F10,6,1H,),1X,4HNOZZ,I3)
520 FORMAT (6X,1H7,6(F10,6,1H,),1X,4HNOZZ,I3)
530 FORMAT (6X,1H8,6(F10,6,1H,),1X,4HNOZZ,I3)
540 FORMAT (6X,1H9,6(F10,6,1H,),1X,4HNOZZ,I3)
550 FORMAT (6X,1H10,6(F10,6,1H,),1X,4HNOZZ,I3)
560 FORMAT (6X,1H11,6(F10,6,1H,),1X,4HNOZZ,I3)
570 FORMAT (6X,1H12,6(F10,6,1H,),1X,4HNOZZ,I3)
580 FORMAT (6X,1H13,6(F10,6,1H,),1X,4HNOZZ,I3)
590 FORMAT (6X,1H14,6(F10,6,1H,),1X,4HNOZZ,I3)
600 FORMAT (6X,1H15,6(F10,6,1H,),1X,4HNOZZ,I3)
610 FORMAT (1HC,12X,7HX(INCH),13X,7HY(INCH))
END

```

SUBROUTINE FRAMB (IWORDL,PLOTS)
LOGICAL PLOTS

FRAMB FOR AVCO IBM 360/75
PRUDUCES IDENTIFICATION FRAME FOR PLOTS AND CLOSES PLOT DATA SET
DIMENSION XNAME(3),XMEMO(2)
DATA XNAME /' ','RADE',',', XMEMO /'NOZF',',IT',', BIN /'.29',/
IWORDL=2
IF (PLOTS) CALL IDFRMV (XNAME,BIN,XMEMO)
RETURN FRAMB2
CALL PLTND
RETURN
END

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SUBROUTINE FRAMES (IWORDL,PLOTS)

FRAMES FOR UNIVAC 1108
DUMMY ROUTINE

ENTRY FRAMB2
IWORDL=1
RETURN
END

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FRU
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FRU
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FRU
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SUBROUTINE GRAPH (NF,LX,LY,NP,X,Y,XL, ,YL,YU,DC,NC,LINE,ORDL,ABSL,GRA
1,TITL,LL,PVAL,PLABL,ICURVE)
DIMENSION X(120),Y(120),LINE(3),CRDL(12),ABSL(12),TITL(12),
1 PLABL(3,2),NXY(2),IMAG(2),XYMAG(2),NC(2)
NOTE - 48 CHARACTERS IN ABSCISSA, ORDNATE, AND TITLE LABELS
NOTE - 12 CHARACTERS IN EACH OF 2 LINES OF PARAMETER LABEL (PLABL)
NOTE - CALL SETMIV (30,125,24,24) BEFORE FIRST CALL TO GRAPH,
IF PARAMETER TABLE IS TO BE PRINTED
IF (NF.EQ.0) GO TO 110
JCURVE=0
NPL=0
CALL SMXYV (LX,LY)
IF (LX.EQ.0) GO TO 10
DX=1.
NN=1
GO TO 20
CALL DXDYV (1,XL,XU,DX,NN,III,NXX,DC,IERR)
DY=1.
MMM=1
GO TO 40
CALL DXDYV (2,YL,YU,DY,MMM,JJJ,NYY,DC,IERR)
IF (IERR.EQ.0) GO TO 50
WRITE (6,310) XL,XU,YL,YU
RETURN
XYMAG(1)=AMAX1(ABS(XL),ABS(XU))
XYMAG(2)=AMAX1(ABS(YL),ABS(YU))
DO 90 I=1,2
IMAG(I)=ALOG10(XYMAG(I))
IF (IMAG(I).GT.5)OR(IMAG(I).LT.-3) GO TO 80
IF (IMAG(I)) 60,60,70
NXY(I)=YABS(IMAG(I))+4
GO TO 90
NXY(I)=IMAG(I)+1
GO TO 90
NXY(I)=3
CONTINUE
III=III
JJJ=JJJ
IF (LL.LT.2) GO TO 100
XL=YL-.022*(XU-XL)
DYL=0.022*(YU-YL)
YL=YL-DYL
YU=YU+DYL
CALL GRIDIV (1,XL,XU,YL,YU,DX,DY,NN,MMM,III,JJJ,NXY(1),NXY(2))
ICURVE=0
ICURVE=ICURVE+1
IF (ICURVE.EQ.1) NPL=NPL+1
JCURVE=JCURVE+1
IXI=NXY(X(1))
IYI=NXY(Y(1))
NCR=NC(2)
NCF=NC(1)
IF (NCF) 120,140,130

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120 NSYMBL=ICURVE
130 GO TO 140
140 NSYMBL=NC1
150 DD 180 I=1,NP
160 IF (NC1.EQ.0) GO TO 150
170 CALL POINTB (IX1,IY1,NSYMBL,NC2)
180 IF (I.EQ.NP) GO TO 180
190 I2=I+1
200 IF (LINE(I)) 170,170,160
210 CALL LINEB (X(I),Y(I),X(I2),Y(I2),LINE)
220 IX1=NXV(X(I2))
230 IY1=NYV(Y(I2))
240 CONTINUE
250 IF (JCURVE.GT.1) GO TO 190
260 CALL CHSIZV (3,3)
270 CALL RITE2V (80,9,1023,90,2,48,1,ABSL,NLAST)
280 CALL RITE2V (9,80,1023,180,2,48,1,ORDL,NLAST)
290 CALL PITE2V (80,1012,1023,90,2,48,1,TITL,NLAST)
300 IF (LL.EQ.0) RETURN
310 CURVE=ICURVE
320 IF (LL.EQ.1) GO TO 200
330 NXV1=NXV(X(1))
340 NYV1=NYV(Y(1))
350 IXV1=NXV(X(2))-NXV1
360 IYV1=NYV(Y(2))-NYV1
370 HYP=IXV1**2+IYV1**2
380 HYP=SQRT(HYP)
390 IX1=NXV1-16.*IDXV/HYP
400 IY1=NYV1-16.*IDYV/HYP
410 CALL LABLV (CURVE,IX1,IY1,2,2,2)
420 IF (ICURVE.GT.1) GO TO 230
430 IY=950-(NPL-1)*300
440 IY2=IY+15
450 CALL LINEBR (930,IY2,1023,IY2,LINE)
460 DD 220 I=1,2
470 DD 210 J=1,2
480 CALL PRINTV (12,PLABL(1,I),924,IY)
490 IY=IY-20
500 IY=IY-20
510 IF (LL.EQ.1) GO TO 240
520 CALL LABLV (CURVE,900,IY,2,1,2)
530 CALL POINTB (924,IY,NSYMBL,NC2)
540 IF (PVAL.EQ.0) GO TO 250
550 IPVM=ALOG10(ABS(PVAL))
560 GO TO 260
570 IPVM=0
580 IF (IPVM.GT.4.OR.IPVM.LT.-2) GO TO 290
590 IF (IPVM) 270,270,280
600 NCHAR=IABS(IPVM)+5
610 NDMAX=1
620 GO TO 300
630 NCHAR=5
640 NDMAX=IPVM+1
650 GO TO 300
660 NDMAX=1

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GRA 57
GRA 58
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GRA 108
GRA 109
GRA 110

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NCHAR=-3
CALL LABLV (PVAL,940,IY,NCHAR,1,NDMAX)
RETURN

GRA 111
GRA 112
GRA 113
GRA 114
GRA 115
GRA 116
GRA 117
GRA 118-

FORMAT (S1H0GRID GENERATION IMPOSSIBLE WITH THE PARAMETERS XL=,1PE
110.3.SH, XU=,E10.3.SH, YL=,E10.3.SH, YU=,E10.3)
END

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SUBROUTINE POINTB (IX,IY,NSYMBL,IDARK)
DIMENSION CHAR(10),K(10)
DATA CHAR /'X','+','.','D','*','H','C','.',',','Y','.',',','H'./
CALL CHSIZV (3,2)
CALL RITE2V (IX,IY,1023,K(NSYMBL),IDARK,1,1,CHAR(NSYMBL),NLAST)
RETURN
END

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POI
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POI

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VY1=VY1+DEL*SINA
NX1=VX1
NY1=VY1
CONTINUE
RETURN
IF (IPAR.EQ.0) GO TO 90
DEL=15.
GO TO 100
DEL=5.
NP=DIS/DEL
DO 120 I=1,NP
DO 110 J=1,IDARK
CALL POINTV (NX1,NY1,-IPAR,1.)
IF (I.EQ.NP) GO TO 120
VX1=VX1+DEL*CSA
VY1=VY1+DEL*SINA
NX1=VX1
NY1=VY1
CONTINUE
RETURN
END

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