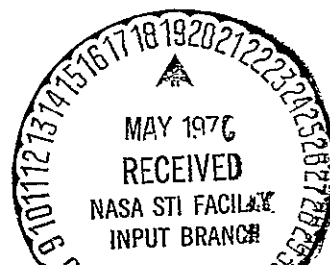


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QUAD ANTENNA SWITCHING EVALUATION
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SHUTTLE ORBITER
S-BAND QUAD ANTENNA
SWITCHING EVALUATION

JANUARY 1975

SPACE SHUTTLE ENGINEERING AND OPERATIONS SUPPORT

NAS 9-13970

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Shuttle Tracking and Communication Support

Antenna and Propagation Evaluation

Design Note

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Prepared by: J. F. Lindsey
J. F. Lindsey
Senior Engineer

Prepared by: David H. Orr
David H. Orr
Associate Engineer

Approved by: G. L. Hornback
for G. L. Hornback
Senior Group Engineer

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ACRONYMS

ATL	Advanced Technology Laboratory
GET	Ground Elapse Time
KSC	Kennedy Space Center
LL	Lower Left
LR	Lower Right
LVLH	Local Vertical/Local Horizontal
MDM	Multiplexer/Demultiplexer
TDRS	Tracking and Data Relay Satellite
TPS	Thermal Protection System
UL	Upper Left
UR	Upper Right
X-POP	X-axis perpendicular to orbit plane

SUMMARY

A study has been performed to evaluate automatic switching of the Shuttle Orbiter S-Band Quad Antennas by the Orbiter on-board computers. The study involved the development and use of an extensive computer program to determine antenna switch position states as a function of time for various orbital activities. The selection of the optimum Quad Antenna element at any given time is based on the look angle to the appropriate Tracking and Data Relay Satellite (TDRS). This study shows that a 2.4 second period is required for updating the S-Band Quad Antenna switch state based on a maximum roll rate of 2° per second. The possibility of a variable update period is suggested since the 2° per second attitude rate is seldom encountered and would, for example, dictate approximately 248,000 on-board computer calculations during Reference Mission 2. The average number of antenna switch state changes was found to be in the range of 1300 for Reference Mission 2.

I. INTRODUCTION

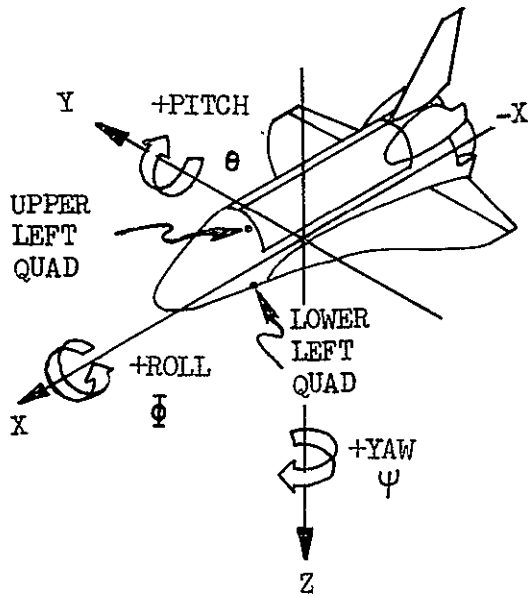
The purpose of this report is to evaluate the antenna switching characteristics of the S-Band Quad Antennas for a typical Shuttle mission. This information is needed in order to assess the required update period(s) for the S-Band Quad Antenna switch from the associated Shuttle Orbiter computer.

Background information is given in Section II in order to explain the operation of the S-Band Quad Antennas, the antenna switch and the associated Shuttle Orbiter computer interfaces. This is followed by Section III which contains a summary of attitude and trajectory profiles of Reference Mission 2 with 3 options. Section IV describes the operation of the ground-based Fortran computer programs used to simulate automatic switching and mission activity, and this is followed by Section V which gives the results of the computer runs which were made at the Johnson Space Center. Finally, in Section VI the results of the computer runs are combined with the trajectory and attitude profiles to evaluate antenna switching characteristics for Reference Mission 2 with a recommendation for the antenna switch update period.

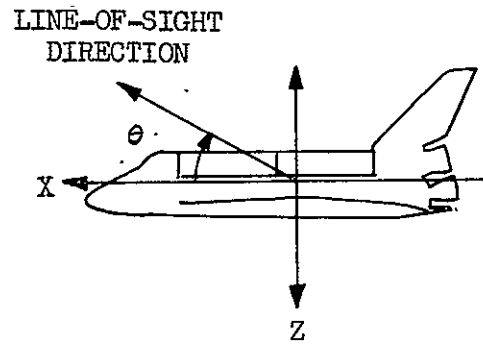
II. BACKGROUND

The S-Band Quads consist of four separate antennas mounted 90° apart near the surface of the Shuttle Orbiter (See Figure 1a.).

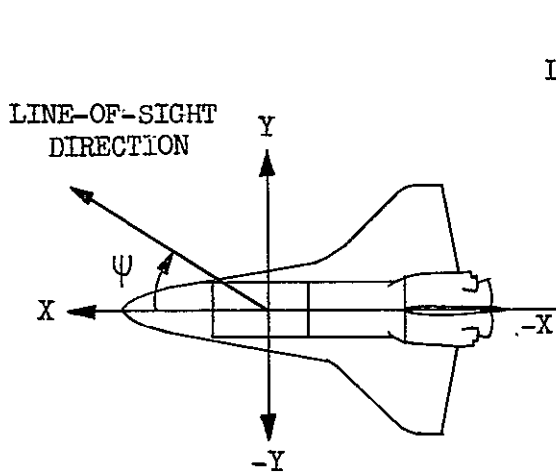
The antennas provide communication between the Shuttle Orbiter and the ground either directly or via a synchronous Tracking and Data Relay Satellite (TDRS). In both cases the S-Band Quad Antennas may be switched separately by manual or automatic selection. In the automatic selection mode real-time calculations are performed by Orbiter on-board computers to determine the "optimum" antenna. This selection is based on the antenna coverage and the line-of-sight look angle between the Shuttle Orbiter and appropriate TDRS or ground station. For the purpose of this study each Quad Antenna was assumed to have a preferred pattern in a particular roll quadrant. It should be noted that the performance of each Quad Antenna is specified with the antenna mounted on a minimum ten (10) wavelength diameter ground plane (Reference 1) which is covered by an appropriate thermal protection system (TPS). Each antenna has a +3dB or better gain defined over a 140° x 90° ellipse-shaped beam. The gain is referenced to a circular isotropic level; however, as was pointed out in Reference 2 the specification does not guarantee a perfect isotropic level since the word "perfect" is omitted. This may be corrected by inserting the word "perfect" in the specification or by specifying the antenna efficiency. The antenna patterns for the Quad Antennas mounted on the Shuttle Orbiter will be somewhat different from those on the circular ground plane due to curvature effects, shadowing and antenna axis canting



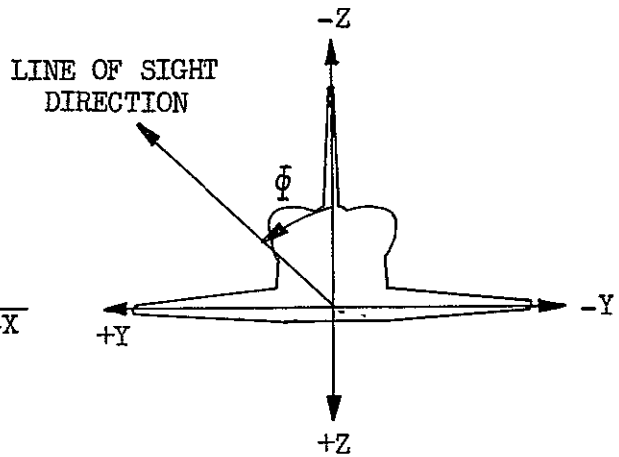
(a) YAW, PITCH, ROLL DIRECTIONS



(b) PITCH ANGLE



(c) YAW ANGLE



(d) ROLL ANGLE

FIGURE 1. SHUTTLE ORBITER BODY COORDINATE SYSTEM

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at each Shuttle Orbiter location. Yet, until accurate patterns are made, it is very reasonable to assume that "optimum" coverage is obtained in the appropriate roll quadrant. For example, if the line-of-sight look angle between the Shuttle Orbiter and a TDRS is in the roll quadrant between the +Y and -Z axes the upper right (UR) antenna is selected. For look angles in the -Z/-Y quadrant the upper left is selected, etc. An illustration of the Shuttle Orbiter coordinate system is found in Reference 3 and is redrawn in Figure 1 to define the roll angle (ϕ), the pitch angle (θ) and the yaw angle (ψ). The reference points for roll, pitch and yaw are not given in Reference 1; therefore, for the purpose of this report the 0° roll angle is referenced to the -Z axis, the 0° pitch is referenced to the +X axis and the 0° yaw is referenced to +X axis. It should be noted that a unique line-of-sight look angle is defined by specifying any two attitude angles and the sequence. For example, a yaw of 30° and a pitch of 30° would correspond to a line-of-sight look angle in the -Z/+Y roll quadrant and the upper right Quad would be selected.

The selection of the proper antenna is made by an Orbiter computer which controls the antenna switch (Reference 4). A simplified functional diagram is shown in Figure 2 to illustrate the interfaces between the antenna switch and the Orbiter computer. The antenna switch has four input ports two of which are connected directly to Transponder #1 and Transponder #2 in the low power mode. The other two input ports are used for the high power mode with a paramplifier for either a high or low S-Band frequency. The four output ports are connected to

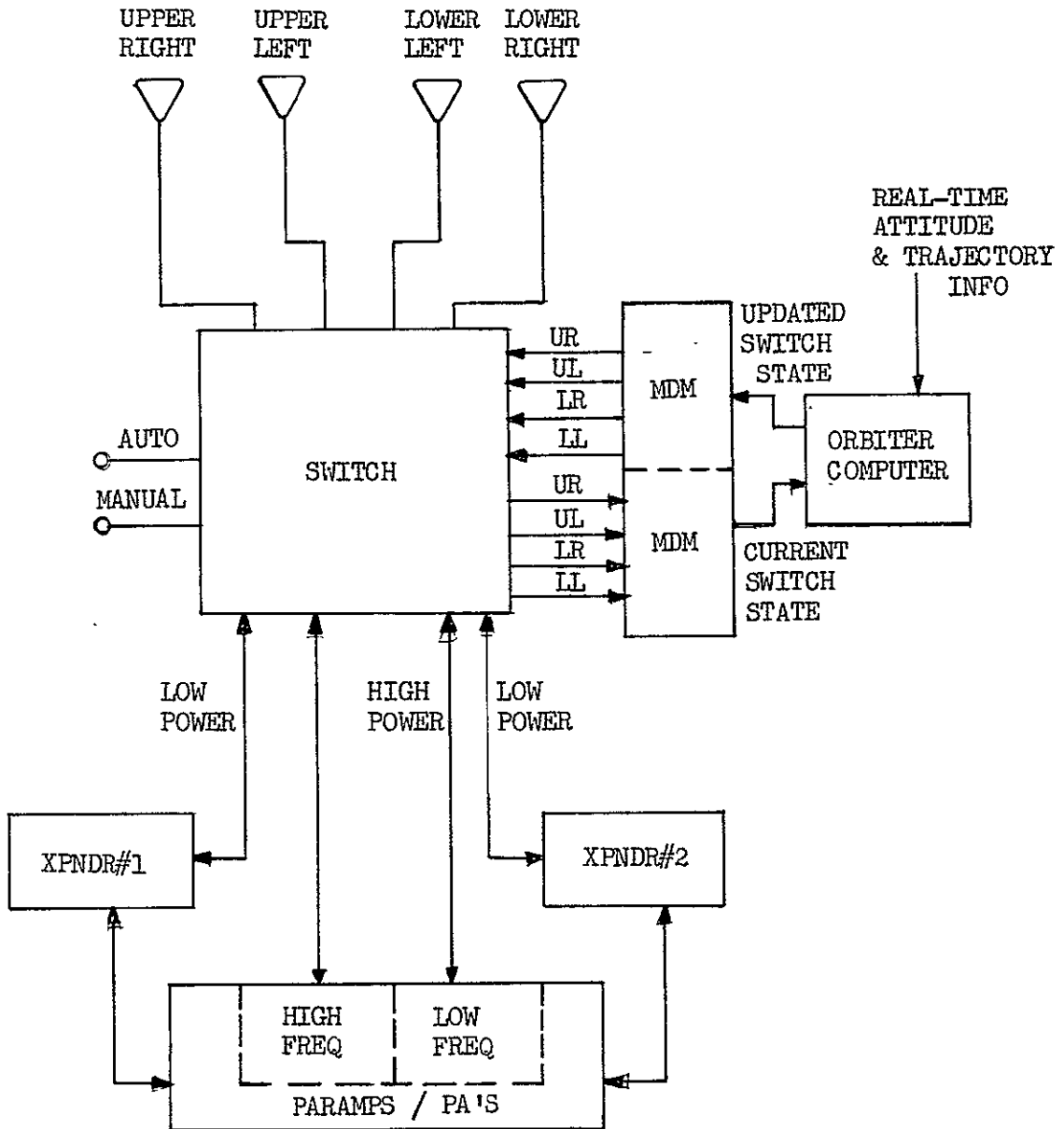


FIGURE 2. S-BAND QUAD ANTENNA SWITCH INTERFACES

individual Quad Antennas. The switch is controlled (Reference 4) so that any one input port and any one output port are connected. In the low power mode, communication is made directly with one of five (5) ground stations and in the high power mode, which is used most of the time, communication is relayed to a ground station via a TDRS. The control of the antenna selection may be manual or automatic via the associated Orbiter computer. The computer interfaces with the switch through the Multiplexer-Demultiplexer (MDM). Signals going to the computer include the present antenna state and the selection of TDRS (high power) or direct ground communication (low power). Also, real-time attitude and trajectory information are inputs to the computer. The signals from the computer to the switch determine the updated antenna switch position (i.e. upper right, upper left, lower right or lower left). The output from the computer may consist of a 2-bit word for determining the four unique positions. The antenna selection is determined by computing the line-of-sight look angle between the Shuttle Orbiter and the appropriate TDRS or ground station. This look angle is then used to determine the "optimum" antenna and an updated signal is sent to the switch through the MDM. The major effort of this study is to assess the required update period for the antenna switch based on Shuttle mission activities.

III. ATTITUDE AND TRAJECTORY PROFILES

Reference Mission 2 is a seven (7) day combined satellite servicing and sortie mission. It is one of three reference missions designed to illustrate current Shuttle operational functions, techniques, and philosophy. Mission 2 is assumed to have three sortie experiment options each representing a major scientific discipline area to be supported by the Orbiter. These three sortie disciplines are astronomy, earth observation and space processing.

For Reference Mission 2 the Shuttle is assumed to be launched from the Kennedy Space Center (KSC) at 9 a.m. on December 1, 1986. During launch the vehicle rolls at a maximum of 9° per second (Reference 5). After the external tank is jettisoned, the orbital maneuver system insertion burn places the Orbiter into a 100-by-50 nautical mile orbit. A maneuver at the first apogee essentially circularizes the orbit, and rendezvous phasing with the satellite is begun. The rendezvous is completed approximately one day later.

After rendezvous and docking with the satellite, refurbishment operation commences at 26 hours ground elapse time (GET). Refurbishment operations terminate at 46 hours GET at which time servicing equipment is stowed. At 49 hours GET the orbit is adjusted to establish a perigee of 230 n. mi. and then a circularization maneuver is performed at perigee 48 minutes later. At 50 hours GET the Orbiter is oriented to the attitude required for the sortie operations. Sortie operations are performed

during the next 5 days. At 161 hours 30 minutes payload closeout operations begin and preparations for deorbit commence. The deorbit is executed at 164 hours 28 minutes GET with the landing at KSC at 165 hours 32 minutes and 2 seconds GET.

As previously described Mission 2 can be tailored to support a particular scientific discipline by one of three sortie options. Option 1 is a satellite servicing/astronomy mission, Option 2 is a satellite servicing/earth observation mission, and Option 3 is a satellite servicing/space processing mission:

Option 1 consists of astronomy sorties using the 1.5 meter IR telescope. The Orbiter should be stabilized in an inertial attitude with the Orbiter -Z-axis pointed to the target. The Orbiter orientation about the Z-axis is unconstrained; however, the Orbiter X-axis is assumed to be in the orbital plane (X-IOP).

Option 2 consists of various experiments utilizing the Advanced Technology Laboratory (ATL). The best attitude time line for this payload is a heads down local vertical/local horizontal (LVLH) attitude for long periods of time. Such an attitude requires thermal conditioning to meet TPS constraints.

Space processing comprises the Option 3 time line with all experiments located in the sortie lab. The best attitude seems to be an X-POP

(X-axis perpendicular to the orbital plane) LVLH inertial attitude. The X-axis is reversed every eight hours to comply with thermal conditioning constraints.

Mission 2 time lines were studied to determine the amount of time devoted to specific Orbiter attitude activities. The studied activities include holding inertial attitudes, barbecue rolls for thermal conditioning, platform alignments and attitude realignments. Reference Mission 2 time lines for Options 1, 2, and 3 were obtained from a MDTSCO report entitled "Preliminary Baseline Reference Mission 2 Attitude and Pointing Time Lines" (Reference 6). The times of each activity and the percentage of the mission duration are summarized in Table I. This table does not account for launch and landing. The total mission time for Reference Mission 2 is 165 hours 32 minutes and 2 seconds. Launch takes 9 minutes and 57 seconds. The orbiting period which is used in Table I takes 161 hours 20 minutes and 3 seconds. The landing takes 4 hours 2 minutes and 2 seconds.

The worst case angular rate for orbital operation is given in Reference 7 as 2° per second. The maximum roll rate is given in Reference 5 as 9° per second for launch and the maximum bank angle rate is given as 5° per second for landing.

ORBITER ACTIVITY \ OPTION	OPTION 1	OPTION 2	OPTION 3
INERTIAL HOLD	58.2%	53.8%	70%
SLEEP PERIOD (THERMAL CONDX OR INERTIAL HOLD)	24.8%	24.8%	24.8%
ATTITUDE REALIGNMENTS (MANEUVER)	13.3%	--	--
PLATFORM ALIGNMENTS (STAR TRACKING)	1.2%	1.2%	1.19%
LVLH ATTITUDE HOLD	--	14.1%	--

TABLE I. ATTITUDE ACTIVITY SUMMARY OF MISSION 2

IV. COMPUTER PROGRAM OPERATION

This section contains a description of the computer programs used to simulate automatic switching. First, the Shuttle Attitude and Pointing Timeline (SAPT) Processor is discussed. Second, the required additions and modifications are discussed including specially developed Subroutines SWITCH, TDRSIF, and SWPRNT. A simplified flow chart is given to explain the overall functions of each subroutine.

The Shuttle Attitude and Pointing Timeline Processor is a sophisticated computer program which calls up many subprograms. The processor has the capability to input, store and edit summary attitude descriptor data to allow definition and generation of summary attitude and pointing time lines. The inputs consist primarily of a set of descriptor data cards defining the attitude and pointing requirements for the event being considered (i.e., inertial hold, roll rate, pitch rate, etc.). The program requires an input trajectory tape for the mission to be simulated, an ephemeris tape for a time period including the mission dates, and a tape with the SAPT program and its associated subroutines. The SAPT program without the associated subprograms is given in Appendix I.

Input descriptor card information is given in Figure 3. The input format fields are given in Figure 3a and a sample descriptor is given in Figure 3b. The first input card following the XQT (execute) command card must be a data card defining the base date from which

to measure GET. This card is required only once per program execution. The attitude and pointing event descriptor cards follow the base data card in a format which might include up to four data cards per descriptor (See Figure 3b.). The first card in the event set is the event name card. Seventy-two characters are allowed to describe the event in an alphameric field format (See Figure 3a.). The second data card contains the attitude requirements. The third and fourth cards are necessary to define target pointing requirements, if any. The event sets are arranged chronologically and treated individually during SAPT execution. Typical events which can be handled by SAPT are target look angle calculations and Shuttle attitude simulations. Celestial, ground and special targets may be specified by reference to a target table included in a SAPT related subroutine or by inputting values to a set of parameters which will locate and define the desired target. As an example of the second target definition method, a TDRS may be specified as an elevated ground target by inputting its longitude, latitude and altitude above the earth's surface as shown in Figure 3b. A typical run deck set up is shown in Figure 4. SAPT has been applied to the problem of calculating look angles (a yaw-pitch sequence) to a TDRS while maintaining the Shuttle attitude required for any specified Orbiter activity. Reference Mission 2 trajectories, an ephemeris tape for the period 1986-1995, a modified version of the SAPT processor, and three special subroutines are combined to form the computational basis for calculating look angles to TDRS targets.

The direction determined by the yaw-pitch look angle sequence is converted into a Shuttle roll quadrant to determine the proper S-Band Quad Antenna for communication to a TDRS. (Note: a pitch-yaw sequence may result in a different look angle direction).

Subroutine SWITCH determines the roll quadrant based on each yaw-pitch look angle using the limits shown in Table II. A listing of the Subroutine SWITCH appears in Appendix II.

Look angle calculations are performed at the same GET's for both synchronous TDRS's. Both TDRS's are located 5.6107 earth radii above the equator with one at a longitude of 41°W and the other TDRS at 171°W. Since communication is assumed to be with only one TDRS at any given time, a subroutine TDRSIF was written which defines the range limits of each TDRS. A comparison between the Shuttle longitude and each TDRS longitude determines the closer TDRS for communication. A method for insuring proper TDRS target longitudes is employed in TDRSIF so that consecutive descriptors alternate between TDRS's. A listing of the Subroutine TDRSIF appears in Appendix III.

Subroutine SWPRNT was written as a means for controlling the printed output resulting from TDRS and antenna selections. For example, if the currently selected antenna is **the** same as the antenna previously selected, then no switch change is required. SWPRNT will not allow the event to be printed if the Shuttle longitude is out of the range of the current TDRS longitude limits or if the current selected antenna is the

For $0^\circ < \text{YAW} \leq 180^\circ$		For $180^\circ < \text{YAW} \leq 360^\circ$
Upper Right (UR)	$0^\circ < \text{pitch} \leq 90^\circ$	Upper Left (UL)
Upper Left (UL)	$90^\circ < \text{pitch} \leq 180^\circ$	Upper Right (UR)
Lower Left (LL)	$180^\circ < \text{pitch} \leq 270^\circ$	Lower Right (LR)
Lower Right (LR)	$270^\circ < \text{pitch} \leq 360^\circ$	Lower Left (LL)

TABLE II. LIMITS FOR SUBROUTINE SWITCH

same as the previously selected antenna. A listing of SWPRNT is included in Appendix IV.

A simplified flowchart of the SAPT processor along with the three special subroutines is shown in Figure 5. The descriptor input event set is read by a subroutine called in SAPT. The event set defines the attitude for the Shuttle and the target information, which in this case is the TDRS located at 41°W longitude. A yaw-pitch look angle to the TDRS is calculated in SAPT. However, if the Shuttle is not in the range of the TDRS 41°W, then the look angles to this target will be disregarded and look angles to TDRS 171°W will be computed. Descriptors are arranged so that calculations to both TDRS's are computed at simultaneous GET's. If the test in TDRSIF is successful, that is the Shuttle is closer to the TDRS at 41°W, then an antenna selection is made in Subroutine SWITCH. SWITCH also compares the currently selected antenna to the previously selected antenna. When a different antenna selection occurs an output is to be printed. Otherwise, no information is printed and the next input event is read. Subroutine SWPRNT is then called by SAPT assuming an antenna change has occurred. SWPRNT prints the selected antenna and the current number of total switch position changes. The next event set is then read. A printout of an actual run deck used for simulation of a 0.5° per second roll is given in Appendix V.

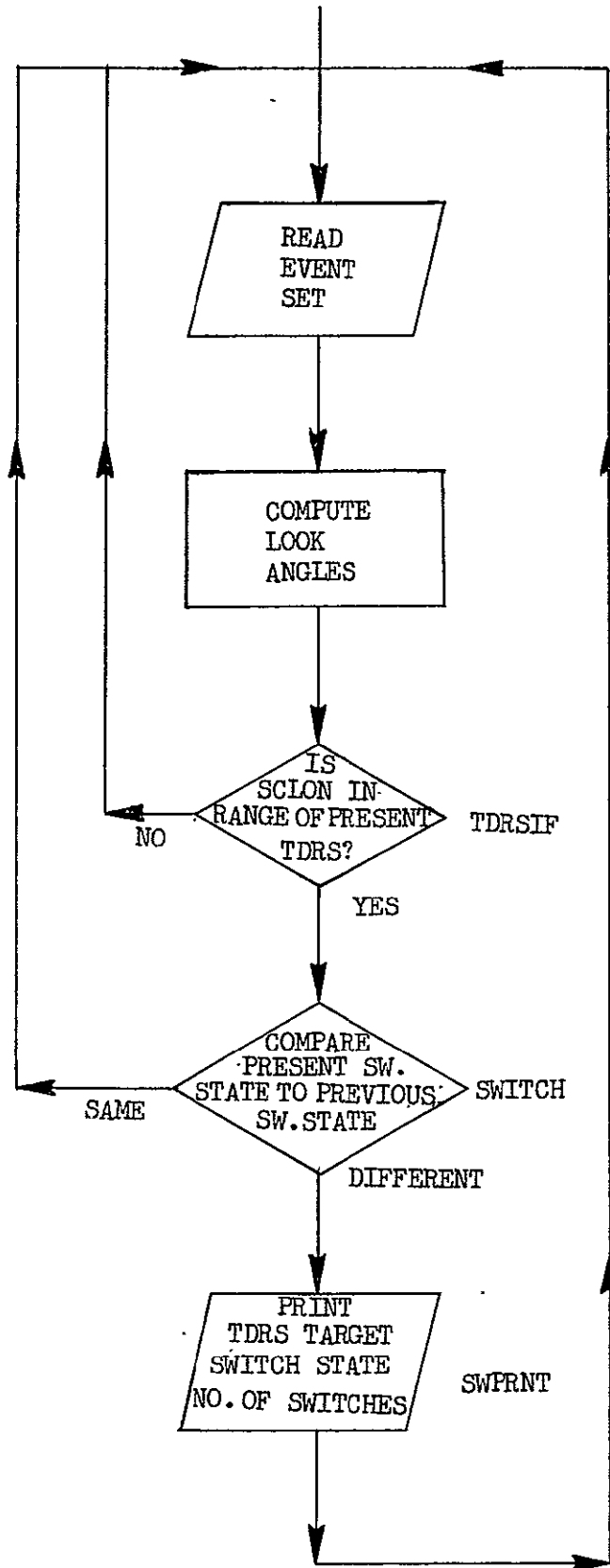


FIGURE 5. FLOW CHART FOR SAPT/SWITCH/SWPRNT/TDRSIF

V. COMPUTER PROGRAM RESULTS

The results of the computer runs are shown in Tables II-XI. In each run the GET begins at approximately 50 hours from launch and represents particular mission activities. This approach of simulating particular activities was chosen for two reasons. The first is the presently planned activities may be changed at a future date, thus exact simulation of a particular mission would be invalid at a later date. The second reason is that a detailed simulation could easily require 50 hours of Johnson Space Center computer time which would have an excessive cost. The simulations for each activity last for approximately one orbit except for star tracking which lasts only 60 minutes due to earth blockage effects.

Table III shows that 31 switch position changes are required for an earth centered inertial (ECI) roll rate of 0.5° per second during a single orbit. It is noted that the TDRS at -171° (171° W) longitude is used for the first 42 minutes and that the -41° (41° W) longitude TDRS is used during the remainder of the period. Calculations for this printout were made every minute and a printout of data occurs only when an antenna change occurs. Also, the antenna selected and the number of position changes is printed. For example, at 50 hours 46 minutes 0 seconds the lower right antenna is used with the TDRS at 41° W longitude. Three minutes later the antenna is changed to the upper right with the same TDRS.

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET IRS MN SECS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK-ANGLES TO SUN		LOOK-ANGLES TO EARTH		LOOK-ANGLES TO TARGET	
		ALT N MI	LAT. DEG	LON DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG	
50 0	TDRS -171	233.6	-50.5	93.8	.0	-.0	-.0	156.3	16.1	323.4	86.4	345.4	.0	270.0	337.9	347.6
	SBQANT IS LOWER LT				1	SWITCH POS CHANGES										
50 2	TDRS -171	236.7	-53.4	105.0	.0	7.8	60.0	156.3	16.1	23.4	85.1	45.3	98.9	330.3	.5	334.5
	SBQANT IS LOWER RT				2	SWITCH POS CHANGES										
50 5	TDRS -171	237.2	-55.2	123.8	.0	19.3	150.0	156.3	16.1	113.4	274.8	44.5	125.1	54.8	25.8	.8
	SBQANT IS UPPER RT				3	SWITCH POS CHANGES										
50 8	TDRS -171	237.2	-53.8	142.9	.0	31.0	240.0	156.3	16.1	203.4	274.9	314.7	235.3	25.4	359.1	25.9
	SBQANT IS UPPER LT				4	SWITCH POS CHANGES										
50 11	TDRS -171	237.1	-49.4	159.4	.1	42.5	330.0	156.3	16.1	293.4	85.2	315.5	208.6	320.3	334.3	359.4
	SBQANT IS LOWER LT				5	SWITCH POS CHANGES										
50 15	TDRS -171	235.9	-40.4	176.2	.1	58.0	90.1	156.3	16.1	53.4	76.6	75.0	148.0	.0	13.0	338.3
	SBQANT IS LOWER RT				6	SWITCH POS CHANGES										
50 18	TDRS -171	235.9	-32.3	174.3	.2	69.6	180.1	156.3	16.1	143.4	273.6	14.6	180.1	20.4	21.9	10.9
	SBQANT IS UPPER RT				7	SWITCH POS CHANGES										
50 21	TDRS -171	233.8	-23.4	166.5	.5	81.2	270.4	156.3	16.1	233.4	283.4	285.0	188.8	359.9	349.9	21.1
	SBQANT IS UPPER LT				8	SWITCH POS CHANGES										
50 24	TDRS -171	234.1	-14.1	159.8	.1	87.2	178.5	156.3	16.1	323.4	86.4	345.4	179.9	2.8	339.2	352.1
	SBQANT IS LOWER LT				9	SWITCH POS CHANGES										
50 27	TDRS -171	232.1	-4.6	153.7	.1	75.6	269.7	156.3	16.1	53.4	76.6	75.0	194.4	.1	6.6	340.2
	SBQANT IS LOWER RT				10	SWITCH POS CHANGES										
50 30	TDRS -171	232.5	5.0	147.9	.1	64.0	359.8	156.3	16.1	143.4	273.6	14.6	180.1	334.0	19.0	4.7
	SBQANT IS UPPER RT				11	SWITCH POS CHANGES										
50 33	TDRS -171	232.3	14.5	141.8	.1	52.4	89.8	156.3	16.1	233.4	283.4	285.0	142.4	359.9	356.5	17.9
	SBQANT IS UPPER LT				12	SWITCH POS CHANGES										
50 36	TDRS -171	232.7	23.7	135.1	.1	40.8	179.9	156.3	16.1	323.4	86.5	345.4	179.9	49.2	343.1	357.8
	SBQANT IS LOWER LT				13	SWITCH POS CHANGES										
50 39	TDRS -171	233.6	32.6	127.3	.1	29.2	269.9	156.3	16.1	53.4	76.6	75.0	240.8	.1	1.6	344.3
	SBQANT IS LOWER RT				14	SWITCH POS CHANGES										
50 42	TDRS -171	233.2	40.7	117.7	.1	17.6	359.9	156.3	16.1	143.4	273.5	14.6	180.2	287.6	14.6	1.2
	SBQANT IS UPPER RT				15	SWITCH POS CHANGES										
50 45	TDRS -41	236.4	47.6	105.5	.1	6.0	89.9	156.3	16.1	233.4	283.4	285.0	96.0	359.9	198.4	309.1
	SBQANT IS LOWER LT				16	SWITCH POS CHANGES										
50 46	TDRS -41	236.4	49.6	100.7	.1	2.2	120.0	156.3	16.1	263.4	77.3	285.8	92.5	29.9	159.8	309.3
	SBQANT IS LOWER RT				17	SWITCH POS CHANGES										
50 49	TDRS -41	236.2	53.9	84.1	.1	350.6	210.0	156.3	16.1	353.4	86.4	15.4	288.4	58.7	126.7	13.0
	SBQANT IS UPPER RT				18	SWITCH POS CHANGES										
50 52	TDRS -41	237.5	55.2	65.0	.1	339.0	300.0	156.3	16.1	83.4	282.7	74.2	293.9	332.2	201.6	52.3
	SBQANT IS UPPER LT				19	SWITCH POS CHANGES										
50 55	TDRS -41	236.6	53.3	46.2	.1	327.4	30.0	156.3	16.1	173.4	273.6	344.6	38.0	313.2	235.3	347.4
	SBQANT IS LOWER LT				20	SWITCH POS CHANGES										
50 58	TDRS -41	236.1	48.5	30.2	.1	315.8	120.1	156.3	16.1	263.4	77.3	285.8	41.6	21.1	159.3	305.6
	SBQANT IS LOWER RT				21	SWITCH POS CHANGES										
51 1	TDRS -41	235.0	41.0	17.5	.1	304.2	210.2	156.3	16.1	353.4	86.5	15.4	341.1	29.1	123.0	10.6
	SBQANT IS UPPER RT				22	SWITCH POS CHANGES										
51 4	TDRS -41	235.3	33.8	7.6	.1	292.6	300.3	156.3	16.1	83.4	282.7	74.2	340.3	348.8	197.2	56.3
	SBQANT IS UPPER LT				23	SWITCH POS CHANGES										
51 7	TDRS -41	235.9	25.1	.4	.1	281.0	30.8	156.3	16.1	173.4	273.5	344.6	5.7	350.6	237.8	352.1
	SBQANT IS LOWER LT				24	SWITCH POS CHANGES										
51 10	TDRS -41	231.3	15.8	7.2	.1	270.7	283.9	156.3	16.1	263.4	77.4	285.8	359.4	359.8	168.4	302.9
	SBQANT IS LOWER RT				25	SWITCH POS CHANGES										
51 13	TDRS -41	230.8	6.3	13.4	.1	282.2	29.1	156.3	16.1	353.4	86.5	15.4	6.0	349.3	122.8	4.7
	SBQANT IS UPPER RT				26	SWITCH POS CHANGES										

Table III. ECI Roll of 0.5° per second Simulation

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MN SECS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
		ALT N MI	LAT DEG	LOX DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
51 16 .00	TDRS -41	230.1	-3.3	19.3	.5	293.8	119.5	156.3	16.1	83.4	282.6	74.2	21.0	11.5		185.7	56.3
	SBQANT IS UPPER LT				27	SWITCH POS CHANGES											
51 19 .00	TDRS -41	230.7	-12.0	25.3	.3	305.5	209.6	156.3	16.1	173.4	273.5	344.6	340.6	30.3		235.4	358.1
	SBQANT IS LOWER LT				28	SWITCH POS CHANGES											
51 22 .00	TDRS -41	231.3	-22.1	31.9	.3	317.1	299.7	156.3	16.1	263.4	77.4	285.8	317.0	338.7		178.7	306.0
	SBQANT IS LOWER RT				29	SWITCH POS CHANGES											
51 26 .00	TDRS -41	232.0	-33.9	42.3	.2	332.5	59.8	156.3	16.1	23.4	85.2	45.3	59.0	333.5		132.2	22.9
	SBQANT IS UPPER RT				30	SWITCH POS CHANGES											
51 29 .00	TDRS -41	233.2	-41.9	52.3	.2	344.2	149.9	156.3	16.1	113.4	274.8	44.5	60.5	56.3		210.2	41.9
	SBQANT IS UPPER LT				31	SWITCH POS CHANGES											

Table III. (Continued)

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MN SECS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
		ALT N MI	LAT DEG	LOX DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
50 0 .0	TDRS -171	233.6	-50.5	93.8	.0	-0.0	-0.0	156.3	16.1	323.4	86.4	345.4	.0	270.0		337.9	347.6
	SBQANT IS LOWER LT				1	SWITCH POS CHANGES											
50 6 .0	TDRS -171	237.2	-55.1	130.3	180.0	336.8	180.0	336.3	343.9	216.6	93.6	14.6	.0	66.8		202.4	13.3
	SBQANT IS UPPER LT				2	SWITCH POS CHANGES											
50 12 .0	TDRS -171	237.0	-47.4	164.1	.1	46.4	.0	156.3	16.1	323.4	86.4	345.4	180.0	316.4		337.6	347.2
	SBQANT IS LOWER LT				3	SWITCH POS CHANGES											
50 18 .0	TDRS -171	235.9	-32.3	174.3	180.2	290.4	179.9	336.3	343.9	216.6	93.6	14.6	.1	20.4		201.9	10.9
	SBQANT IS UPPER LT				4	SWITCH POS CHANGES											
50 24 .0	TDRS -171	234.1	-14.1	159.8	178.6	87.2	178.5	156.3	16.1	323.4	86.4	345.4	179.9	2.8		339.2	352.1
	SBQANT IS LOWER LT				5	SWITCH POS CHANGES											
50 30 .0	TDRS -171	232.5	5.0	147.9	359.8	296.0	.2	336.3	343.9	216.6	93.6	14.6	.1	334.0		199.0	4.7
	SBQANT IS UPPER LT				6	SWITCH POS CHANGES											
50 36 .0	TDRS -171	232.7	23.7	135.1	179.9	40.8	179.9	156.3	16.1	323.4	86.5	345.4	179.9	49.2		343.1	357.8
	SBQANT IS LOWER LT				7	SWITCH POS CHANGES											
50 42 .0	TDRS -171	233.2	40.7	117.7	359.9	342.4	.1	336.3	343.9	216.6	93.5	14.6	.2	287.6		194.6	1.2
	SBQANT IS UPPER LT				8	SWITCH POS CHANGES											
50 45 .0	TDRS -41	236.4	47.6	105.5	.4	84.0	.5	225.7	309.5	295.9	75.4	3.4	179.9	354.0		75.6	323.2
	SBQANT IS LOWER RT				9	SWITCH POS CHANGES											
50 49 .0	TDRS -41	236.2	53.9	-84.1	179.9	320.6	180.0	133.8	38.8	312.8	94.4	345.7	.0	50.6		127.3	7.0
	SBQANT IS UPPER RT				10	SWITCH POS CHANGES											
50 55 .0	TDRS -41	236.6	53.3	-46.2	359.8	62.6	359.9	313.8	321.2	227.2	85.6	14.3	180.0	332.6		53.9	354.6
	SBQANT IS LOWER RT				11	SWITCH POS CHANGES											
51 1 .0	TDRS -41	235.0	41.8	-17.5	178.4	274.2	181.4	133.8	38.8	312.8	94.4	345.7	359.9	4.2		126.1	3.0
	SBQANT IS UPPER RT				12	SWITCH POS CHANGES											
51 7 .0	TDRS -41	235.9	25.1	.4	180.3	71.0	180.5	313.8	321.2	227.2	85.6	14.3	180.2	19.0		52.6	359.7
	SBQANT IS LOWER RT				13	SWITCH POS CHANGES											
51 14 .0	TDRS -41	230.2	3.1	15.3	.0	346.1	359.8	95.3	53.8	284.2	101.0	349.8	359.2	283.9		127.3	17.0
	SBQANT IS UPPER RT				14	SWITCH POS CHANGES											
51 20 .0	TDRS -41	230.7	-15.9	27.4	180.0	350.7	180.2	275.3	306.2	255.8	79.0	10.2	358.7	80.7		49.2	342.3
	SBQANT IS LOWER RT				15	SWITCH POS CHANGES											
51 26 .0	TDRS -41	232.8	-33.9	42.3	.0	32.5	359.8	95.3	53.8	284.2	101.0	349.8	180.3	302.5		133.8	19.7
	SBQANT IS UPPER RT				16	SWITCH POS CHANGES											

Table IV. ECI pitch of .5° per second simulation

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MN SECS	EVENT	VEHICLE POSITION			LVH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET			
		ALT N MI	LAT DEG	LOX DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG	
50 0	TDRS -171	233.6	-50.5	93.8	0	-0	-0	156.3	16.1	323.4	86.4	345.4	0	270.0		337.9	347.6	
	SUBQANT IS LOWER LT				1 SWITCH POS CHANGES													
50 6	TDRS -171	237.2	-55.1	130.3	180.0	336.8	-0	336.3	343.9	36.6	266.4	345.4	360.0	293.2		157.6	346.7	
	SUBQANT IS LOWER RT				2 SWITCH POS CHANGES													
50 12	TDRS -171	237.0	-47.4	164.1	0	46.4	0	156.3	16.1	323.4	86.4	345.4	180.0	316.4		337.6	347.2	
	SUBQANT IS LOWER LT				3 SWITCH POS CHANGES													
50 18	TDRS -171	235.9	-32.3	-174.3	180.2	290.4	359.9	336.3	343.9	36.6	266.4	345.4	359.9	339.6		158.1	349.1	
	SUBQANT IS LOWER RT				4 SWITCH POS CHANGES													
50 24	TDRS -171	234.1	-14.1	-159.8	178.6	87.2	178.5	156.3	16.1	323.4	86.4	345.4	179.9	2.8		339.2	352.1	
	SUBQANT IS LOWER LT				5 SWITCH POS CHANGES													
50 30	TDRS -171	232.5	5.0	-147.9	359.8	296.0	180.2	336.3	343.9	36.6	266.4	345.4	359.9	26.0		161.0	355.3	
	SUBQANT IS LOWER RT				6 SWITCH POS CHANGES													
50 36	TDRS -171	232.7	23.7	-135.1	179.9	40.8	179.9	156.3	16.1	323.4	86.5	345.4	179.9	49.2		343.1	357.8	
	SUBQANT IS LOWER LT				7 SWITCH POS CHANGES													
50 42	TDRS -171	233.2	40.7	-117.7	359.9	342.4	180.1	336.3	343.9	36.6	266.5	345.4	359.8	72.4		165.4	358.8	
	SUBQANT IS LOWER RT				8 SWITCH POS CHANGES													
50 45	TDRS -41	236.4	47.6	-105.5	269.9	0	186.0	77.9	325.1	340.2	176.5	345.4	269.5	84.0		217.6	348.5	
	SUBQANT IS LOWER LT				9 SWITCH POS CHANGES													
50 47	TDRS -41	236.4	51.3	-95.6	209.9	358.6	179.1	132.6	357.4	320.6	116.5	345.4	31.3	88.3		157.0	348.7	
	SUBQANT IS LOWER RT				10 SWITCH POS CHANGES													
50 53	TDRS -41	237.2	54.9	-88.6	32.3	21.4	193.0	312.6	2.6	39.4	296.5	345.4	210.0	65.1		335.2	348.3	
	SUBQANT IS LOWER LT				11 SWITCH POS CHANGES													
50 59	TDRS -41	235.7	46.5	-25.6	220.6	319.8	151.0	132.6	357.4	320.6	116.5	345.5	29.9	41.9		153.8	346.4	
	SUBQANT IS LOWER RT				12 SWITCH POS CHANGES													
51 5	TDRS -41	235.4	31.0	-4.8	60.7	55.2	235.7	312.6	2.6	39.4	296.5	345.5	209.8	18.7		333.5	343.6	
	SUBQANT IS LOWER LT				13 SWITCH POS CHANGES													
51 11	TDRS -41	231.0	12.7	9.3	277.7	300.1	81.0	132.6	357.4	320.6	116.5	345.5	29.8	355.5		154.5	340.5	
	SUBQANT IS LOWER RT				14 SWITCH POS CHANGES													
51 17	TDRS -41	230.2	-6.4	21.3	129.0	50.2	316.6	312.6	2.6	39.4	296.5	345.5	209.8	332.3		336.9	338.1	
	SUBQANT IS LOWER LT				15 SWITCH POS CHANGES													
51 23	TDRS -41	231.7	-25.1	34.3	323.6	326.8	21.9	132.6	357.4	320.6	116.5	345.5	29.7	309.1		160.3	336.8	
	SUBQANT IS LOWER RT				16 SWITCH POS CHANGES													
51 29	TDRS -41	233.2	-41.9	52.3	149.2	13.8	352.0	312.6	2.6	39.4	296.5	345.5	209.5	285.8		344.3	336.9	
	SUBQANT IS LOWER LT				17 SWITCH POS CHANGES													

Table V. ECI Yaw of 0.5° per second simulation

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR.

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SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MN SECS	EVENT	VEHICLE POSITION			LVLM ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
		ALT N 41	LAT DEG	LOX DFG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
50 0 .0	TDRS -171	233.6	-50.5	93.8	180.0	346.5	-75.0	327.3	333.2	114.9	.0	270.0	-76.1	345.4		-148.9	-14.3
	SBQANT IS UPPER RT																
50 3 .0	TDRS -171	236.9	-54.3	111.1	180.0	334.9	-165.0	327.3	333.2	204.9	90.0	360.0	-28.9	61.0		-196.6	30.7
	SBQANT IS UPPER LT																
50 6 .0	TDRS -171	237.2	-55.1	130.3	180.0	323.3	-255.0	327.3	333.2	294.9	34.7	90.0	307.6	12.0		212.1	345.8
	SBQANT IS LOWER LT																
50 9 .0	TDRS -171	237.2	-52.6	148.8	180.1	311.7	345.0	327.3	333.2	24.9	270.0	.0	347.0	-320.0		163.3	329.0
	SBQANT IS LOWER RT																
50 12 .0	TDRS -171	237.0	-47.4	164.1	180.1	300.1	75.0	327.3	333.2	114.9	323.1	270.0	-29.3	352.5		148.4	-14.4
	SBQANT IS UPPER RT																
50 15 .0	TDRS -171	235.9	-40.4	176.2	180.2	288.5	164.9	327.3	333.2	204.9	90.0	360.0	5.0	17.8		196.7	29.7
	SBQANT IS UPPER LT																
50 18 .0	TDRS -171	235.9	-32.3	174.3	180.5	276.9	254.6	327.3	333.2	294.9	39.1	90.0	353.3	1.8		209.5	345.4
	SBQANT IS LOWER LT																
50 21 .0	TDRS -171	233.7	-23.4	166.5	359.3	274.7	165.8	327.3	333.2	24.9	270.0	.0	1.2	-4.5		163.6	333.0
	SBQANT IS LOWER RT																
50 24 .0	TDRS -171	234.1	-14.1	159.8	359.8	286.3	255.3	327.3	333.2	114.9	318.9	270.0	344.2	4.1		153.8	14.4
	SBQANT IS UPPER RT																
50 26 .0	TDRS -171	232.0	-7.8	155.7	359.9	294.0	315.2	327.3	333.2	174.9	90.0	330.0	342.6	343.2		180.5	28.5
	SBQANT IS UPPER LT																
50 29 .0	TDRS -171	232.3	1.8	149.8	359.9	305.6	45.2	327.3	333.2	264.9	90.0	60.0	26.9	335.8		206.8	359.1
	SBQANT IS LOWER LT																
50 32 .0	TDRS -171	232.1	11.3	143.8	359.9	317.2	135.1	327.3	333.2	354.9	270.0	30.0	37.3	31.3		178.7	335.0
	SBQANT IS LOWER RT																
50 35 .0	TDRS -171	232.6	20.7	137.4	359.9	328.8	225.1	327.3	333.2	84.9	270.0	300.0	310.5	37.1		156.6	1.2
	SBQANT IS UPPER RT																
50 38 .0	TDRS -171	233.7	29.7	130.0	359.9	340.4	315.1	327.3	333.2	174.9	90.0	330.0	296.8	318.1		181.1	22.0
	SBQANT IS UPPER LT																
50 41 .0	TDRS -171	233.3	38.1	121.1	359.9	352.0	45.1	327.3	333.2	264.9	90.0	60.0	78.8	315.6		200.9	359.5
	SBQANT IS LOWER LT																
50 45 .0	TDRS -41	236.4	47.6	105.5	359.9	7.4	165.1	327.3	333.2	24.9	270.0	.0	116.9	73.3		309.5	8.5
	SBQANT IS UPPER LT																
50 46 .0	TDRS -41	236.4	49.6	100.7	359.9	11.3	195.1	327.3	333.2	54.9	270.0	330.0	232.4	71.2		310.2	345.3
	SBQANT IS LOWER LT																
50 49 .0	TDRS -41	236.2	53.9	84.1	359.9	22.9	285.0	327.3	333.2	144.9	89.9	300.0	246.4	346.2		22.8	311.6
	SBQANT IS LOWER RT																
50 52 .0	TDRS -41	237.5	55.2	65.7	359.8	34.5	15.0	327.3	333.2	234.9	90.0	30.0	159.4	307.3		51.6	15.5
	SBQANT IS UPPER RT																
50 55 .0	TDRS -41	236.7	53.3	46.2	359.8	46.1	105.0	327.3	333.2	324.9	270.1	60.0	137.1	10.3		334.5	49.2
	SBQANT IS UPPER LT																
50 58 .0	TDRS -41	236.1	48.5	30.2	359.7	57.7	194.9	327.3	333.2	54.9	270.0	330.0	189.2	31.1		307.7	342.5
	SBQANT IS LOWER LT																
51 1 .0	TDRS -41	235.0	41.8	17.5	359.6	69.3	284.7	327.3	333.2	144.9	89.9	300.0	200.1	354.8		29.1	311.5
	SBQANT IS LOWER RT																
51 4 .0	TDRS -41	235.3	33.8	7.6	359.0	80.9	14.2	327.3	333.2	234.9	90.0	30.0	177.8	351.2		51.9	20.2
	SBQANT IS UPPER RT																
51 7 .0	TDRS -41	235.9	25.1	.4	183.8	87.5	288.7	327.3	333.2	324.9	270.1	60.0	182.3	359.2		327.8	46.4
	SBQANT IS UPPER LT																
51 10 .0	TDRS -41	231.3	15.8	7.2	180.6	75.9	15.8	327.3	333.2	54.9	270.0	330.0	176.1	346.4		310.0	337.2
	SBQANT IS LOWER LT																
51 13 .0	TDRS -41	230.8	6.3	13.4	180.4	64.3	105.5	327.3	333.2	144.9	89.9	300.0	155.1	6.6		33.7	316.7
	SBQANT IS LOWER RT																

Table VI. Thermal Conditioning - Roll 0.5° per second, X-axis sun simulation

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SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MN SECS	EVENT	VEHICLE POSITION			LVCH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
		ALT N MI	LAT DEG	LOX DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
51 15 .0	TDRS -41	230.1	-7.1	174.3	180.3	56.5	165.4	327.3	333.2	264.9	90.0	360.0	170.5	32.2		51.9	1.2
	SBQANT IS UPPER RT																
51 18 .0	TDRS -41	230.4	-7.6	237.3	180.2	44.9	255.3	327.3	333.2	294.9	39.4	89.9	224.1	10.3		356.5	50.5
	SBQANT IS UPPER LT																
51 21 .0	TDRS -41	231.0	-19.0	295.6	180.2	33.3	345.3	327.3	333.2	24.9	270.0	0.0	201.2	306.1		311.0	357.1
	SBQANT IS LOWER LT																
51 24 .0	TDRS -41	232.1	-28.1	36.8	180.2	21.7	75.2	327.3	333.2	114.9	321.7	270.1	112.4	346.3		4.8	312.9
	SBQANT IS LOWER RT																
51 27 .0	TDRS -41	233.0	-36.7	45.4	180.2	10.1	165.2	327.3	333.2	204.9	90.0	360.0	124.9	72.1		45.4	3.2
	SBQANT IS UPPER RT																
51 30 .0	TDRS -41	233.3	-44.3	56.1	180.2	358.5	255.1	327.3	333.2	294.9	37.5	89.9	271.6	14.9		356.0	43.5
	SBQANT IS UPPER LT																

Table VI (Continued)

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MN SECS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
		ALT N MI	LAT DEG	Lon DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
49 59 .0	ASTRONOMY 520 SBQANT IS UPPER RT	233.8	-48.7	88.8	-0.0	1.0	219.3	153.2	20.0	181.8	272.5	335.6	268.4	50.7		.0	90.0
50 45 .0	TDRS -41 SBQANT IS LOWER LT	236.4	47.6	-105.5	179.9	1.2	39.2	153.2	20.0	181.8	272.4	335.5	91.9	309.3		228.5	337.8

Table VII. Star Tracking - 520 Simulation

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MN SECS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
		ALT N MI	LAT DEG	LOX DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
49 59	•0 ASTRONOMY 522	233.6	-48.7	88.8	-0	328.5	310.6	172.4	353.6	275.5	70.5	300.7	308.9	326.3		0	90.0
	SBQANT IS UPPER RT				1	SWITCH POS CHANGES											
50 0	•0 TDRS -171	233.6	-50.5	93.8	0	332.4	310.6	172.4	353.6	275.5	70.5	300.7	309.6	324.8		355.9	25.4
	SBQANT IS UPPER LT				2	SWITCH POS CHANGES											
50 24	•0 TDRS -171	234.1	-14.1	-159.8	2	65.2	310.8	172.4	353.6	275.5	70.5	300.7	199.3	344.1		3	27.9
	SBQANT IS UPPER RT				3	SWITCH POS CHANGES											
50 45	•0 TDRS -41	236.4	47.6	-105.5	179.9	33.6	130.5	172.4	353.6	275.5	70.6	300.6	131.2	32.8		159.9	297.8
	SBQANT IS LOWER RT				4	SWITCH POS CHANGES											

Table VIII. Star Tracking - 522 Simulation

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS	HN	SCCS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
				ALT N MI	LAT DEG	LOX DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
49	59	.0	ASTRONOMY 530	233.8	-48.7	88.8	0.0	298.7	148.0	191.7	329.9	106.6	287.5	35.7	16.1	24.0		0	90.0
			SBQANT IS UPPER RT																
			1 SWITCH POS CHANGES																
50	0	.0	TDRS -171	233.6	-50.5	93.8	0.0	302.5	148.0	191.7	329.9	106.6	287.5	35.7	18.7	27.1		357.4	311.9
			SBQANT IS LOWER LT																
			2 SWITCH POS CHANGES																
50	45	.0	TDRS -41	236.4	47.6	-105.5	179.8	63.5	327.9	191.7	329.9	106.6	287.5	35.7	194.8	337.8		245.3	68.4
			SBQANT IS UPPER LT																
			3 SWITCH POS CHANGES																

Table IX. Star Tracking - 530 Simulation

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MIN SECS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			LCI ATTITUDE			LOOK-ANGLES- TO SUN		LOOK-ANGLES- TO EARTH		LOOK-ANGLES- TO TARGET		
		ALT N MI	LAT DEG	LONG DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
50 0 00	TDRS -171 SUBANT IS LOWER LT	233.6	-60.5	93.4	0	0	0	156.3	16.1	323.4	86.4	345.4	0	270.0		337.9	347.6
50 45 00	TDRS -91 SUBANT IS LOWER RT	236.4	47.6	-105.4	179.9	6.0	179.9	156.3	16.1	323.4	86.5	345.4	179.5	84.0		127.6	348.5

Table X. Inertial Hold Simulation

SHUTTLE ATTITUDE AND POINTING TIME LINE

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GET HRS MN SECS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK-ANGLES TO SUN		LOOK-ANGLE TO EARTH		LOOK-ANGLE TO TARGET	
		ALT M NI	LAT DEG	LONG DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG
50 0 00	TORS -171 SHUTTLE IS LOWER LT	233.6	-50.5	93.0	0.0	0.0	0.0	156.3	16.1	323.4	86.4	345.4	0.0	270.0	337.9	347.6
					1 SWITCH POS CHANGES											
50 45 00	TORS -41 SHUTTLE IS LOWER RT	236.4	47.6	-105.2	179.4	6.0	179.9	156.3	16.1	323.4	86.5	345.4	179.5	84.0	127.6	348.5
					2 SWITCH POS CHANGES											
51 34 00	TORS -171 SHUTTLE IS LOWER LT	256.0	-52.0	75.2	0.2	3.5	359.9	156.3	16.1	323.4	86.5	345.5	181.2	273.5	356.7	333.6
					3 SWITCH POS CHANGES											

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

Table XI. Heads Down Simulation

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MN SECS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
		ALT N MI	LAT DEG	LONG DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
50 0 0.1	TURS -171	236.1	-51.1	95.7	.0	.0	.0	157.1	15.1	323.7	86.1	345.5	.0	270.0		337.9	348.8
	SBWANT IS LOWER LT				1 SWITCH POS CHANGES												
50 5 0.1	TURS -171	237.5	-55.2	128.2	.0	.0	.0	168.4	359.3	325.1	81.5	347.6	.0	270.0		338.1	5.9
	SBWANT IS UPPER LT				2 SWITCH POS CHANGES												
50 45 0.2	TURS -41	234.1	48.4	-103.8	.0	.0	.0	333.2	340.2	217.4	92.2	14.7	.0	270.0		51.8	8.6
	SBWANT IS UPPER RT				3 SWITCH POS CHANGES												
51 25 0.0	TURS -41	232.8	-32.1	48.6	.0	.0	.0	133.8	38.6	312.9	94.5	345.9	.0	270.0		136.9	357.9
	SBWANT IS LOWER RT				4 SWITCH POS CHANGES												
51 35 0.1	TURS -171	237.5	-53.7	83.3	.0	.0	.0	161.3	8.9	324.6	84.5	346.2	.0	270.0		357.3	341.9
	SBWANT IS LOWER LT				5 SWITCH POS CHANGES												
51 40 0.1	TURS -171	237.5	-54.4	115.1	.0	.0	.0	172.4	353.0	324.8	80.3	348.8	.0	270.0		358.4	.1
	SBWANT IS UPPER LT				6 SWITCH POS CHANGES												

Table XII. LVLH Simulation

SHUTTLE ATTITUDE AND POINTING TIME LINE

GLT			EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET	
HRS	MM	SECS		ALT N MI	LAT DEG	LONG DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG
50	0	00	TDRS -171	233.6	-50.5	93.8	00	-00	-00	156.3	16.1	323.4	86.4	345.4	00	270.0	337.9	347.6
			SBWANT IS LOWER LT				1	SWITCH POS	CHANGES									
50	0	30.0	TDRS -171	233.5	-51.3	96.5	00	1.9	60.0	156.3	16.1	23.4	85.1	45.3	92.2	330.0	02	334.7
			SBWANT IS LOWER RT				2	SWITCH POS	CHANGES									
50	1	30.0	TDRS -171	236.6	-52.8	102.1	00	5.8	180.0	156.3	16.1	143.4	273.6	14.6	180.0	84.2	22.2	12.7
			SBWANT IS UPPER RT				3	SWITCH POS	CHANGES									
50	2	00	TDRS -171	236.7	-53.4	105.0	00	7.8	240.0	156.3	16.1	203.4	274.9	314.7	261.1	29.7	359.5	25.5
			SBWANT IS UPPER LT				4	SWITCH POS	CHANGES									
50	3	00	TDRS -171	236.9	-54.3	111.1	00	11.6	00	150.3	16.1	323.4	86.4	345.4	180.0	281.6	337.7	347.0
			SBWANT IS LOWER LT				5	SWITCH POS	CHANGES									
50	3	30.0	TDRS -171	237.0	-54.7	114.2	00	13.5	60.0	150.3	16.1	23.4	85.1	45.3	105.5	330.9	07	334.3
			SBWANT IS LOWER RT				6	SWITCH POS	CHANGES									
50	4	30.0	TDRS -171	237.1	-55.1	120.6	00	17.4	180.0	156.3	16.1	143.4	273.6	14.6	180.0	72.6	22.3	13.2
			SBWANT IS UPPER RT				7	SWITCH POS	CHANGES									
50	5	00	TDRS -171	237.2	-55.2	123.8	00	19.3	240.0	150.3	16.1	203.4	274.9	314.7	247.9	28.1	359.1	25.8
			SBWANT IS UPPER LT				8	SWITCH POS	CHANGES									
50	6	00	TDRS -171	237.2	-55.1	130.3	00	23.2	00	150.3	16.1	323.4	86.4	345.4	180.0	293.2	337.6	346.7
			SBWANT IS LOWER LT				9	SWITCH POS	CHANGES									
50	6	30.0	TDRS -171	237.2	-54.9	133.5	00	25.1	60.0	156.3	16.1	23.4	85.1	45.3	118.4	333.1	09	334.1
			SBWANT IS LOWER RT				10	SWITCH POS	CHANGES									
50	7	30.0	TDRS -171	237.2	-54.2	139.8	00	29.0	180.0	156.3	16.1	143.4	273.6	14.6	180.0	61.0	22.4	13.4
			SBWANT IS UPPER RT				11	SWITCH POS	CHANGES									
50	8	00	TDRS -171	237.2	-53.8	142.9	00	31.0	240.0	150.3	16.1	203.4	274.9	314.7	235.3	25.4	359.1	25.9
			SBWANT IS UPPER LT				12	SWITCH POS	CHANGES									
50	9	00	TDRS -171	237.2	-52.0	148.8	01	34.8	00	156.3	16.1	323.4	86.4	345.4	180.0	304.8	337.6	346.7
			SBWANT IS LOWER LT				13	SWITCH POS	CHANGES									
50	9	30.0	TDRS -171	237.2	-51.9	151.6	01	36.7	60.0	150.3	16.1	23.4	85.1	45.3	130.8	336.4	08	334.2
			SBWANT IS LOWER RT				14	SWITCH POS	CHANGES									
50	10	30.0	TDRS -171	237.2	-50.3	156.9	01	40.6	180.0	156.3	16.1	143.4	273.6	14.6	180.0	49.4	22.4	13.1
			SBWANT IS UPPER RT				15	SWITCH POS	CHANGES									
50	11	00	TDRS -171	237.1	-49.4	159.4	01	42.5	240.0	156.3	16.1	203.4	274.9	314.7	223.4	21.6	359.4	25.7
			SBWANT IS UPPER LT				16	SWITCH POS	CHANGES									

Table XIII. ECI Roll of 2^o Second Simulation

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MN SECS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
		ALT N MI	LAT DEG	LONG DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
50 12 00	TDRS -171	237.0	+47.4	164.1	01	46.4	00	156.3	16.1	323.4	86.4	345.4	180.0	316.4		337.6	347.2
	SBWANT IS LOWER LT					17 SWITCH POS CHANGES											
50 12 30.0	TDRS -171	236.9	-46.4	166.4	01	48.3	60.0	156.3	16.1	23.4	85.1	45.3	142.4	340.6		03	334.4
	SBWANT IS LOWER RT					18 SWITCH POS CHANGES											
50 13 30.0	TDRS -171	235.9	-44.1	170.6	01	52.2	180.0	156.3	16.1	143.4	273.6	14.6	180.0	37.8		22.3	12.5
	SBWANT IS UPPER RT					19 SWITCH POS CHANGES											
50 14 30.0	TDRS -171	236.0	-41.7	174.4	01	56.1	300.1	156.3	16.1	263.4	77.2	285.8	210.2	343.8		337.9	12.4
	SBWANT IS UPPER LT					20 SWITCH POS CHANGES											
50 15 00	TDRS -171	235.9	-40.4	176.2	01	58.0	01	156.3	16.1	323.4	86.4	345.4	180.0	328.0		337.8	348.0
	SBWANT IS LOWER LT					21 SWITCH POS CHANGES											
50 16 00	TDRS -171	235.9	-37.8	179.6	01	61.9	120.1	156.3	16.1	83.4	282.8	74.2	155.2	13.7		21.6	347.4
	SBWANT IS LOWER RT					22 SWITCH POS CHANGES											
50 16 30.0	TDRS -171	235.9	-36.5	178.7	01	63.8	180.1	156.3	16.1	143.4	273.6	14.6	180.0	26.2		22.1	11.45
	SBWANT IS UPPER RT					23 SWITCH POS CHANGES											
50 17 30.0	TDRS -171	235.9	-33.7	175.7	02	67.6	300.1	156.3	16.1	263.4	77.2	285.8	199.6	349.0		339.0	12.8
	SBWANT IS UPPER LT					24 SWITCH POS CHANGES											
50 18 00	TDRS -171	235.9	-32.3	174.3	02	69.6	01	156.3	16.1	323.4	86.4	345.4	179.9	339.6		338.1	349.1
	SBWANT IS LOWER LT					25 SWITCH POS CHANGES											
50 19 00	TDRS -171	235.8	-29.4	171.5	02	73.4	120.2	156.3	16.1	83.4	282.8	74.2	165.6	8.2		20.4	347.0
	SBWANT IS LOWER RT					26 SWITCH POS CHANGES											
50 19 30.0	TDRS -171	233.7	-27.9	170.2	03	75.4	180.2	156.3	16.1	143.4	273.6	14.6	180.1	14.6		21.7	10.2
	SBWANT IS UPPER RT					27 SWITCH POS CHANGES											
50 20 30.0	TDRS -171	233.8	-24.9	167.7	04	79.3	300.3	156.3	16.1	263.4	77.2	285.8	189.3	354.6		340.4	13.2
	SBWANT IS UPPER LT					28 SWITCH POS CHANGES											
50 21 00	TDRS -171	233.8	-23.4	166.5	05	81.2	04	156.3	16.1	323.4	86.4	345.4	179.9	351.2		338.6	350.6
	SBWANT IS LOWER LT					29 SWITCH POS CHANGES											
50 22 00	TDRS -171	233.9	-20.3	164.2	08	85.1	120.8	156.3	16.1	83.4	282.7	74.2	175.8	2.5		18.8	346.6
	SBWANT IS LOWER RT					30 SWITCH POS CHANGES											
50 22 30.0	TDRS -171	234.0	-16.8	163.1	13	87.0	181.3	156.3	16.1	143.4	273.6	14.6	180.1	3.0		21.1	8.7
	SBWANT IS UPPER RT					31 SWITCH POS CHANGES											
50 23 30.0	TDRS -171	234.1	-15.7	160.9	175.4	89.1	115.3	156.3	16.1	263.4	77.3	285.8	179.2	1.4		342.1	13.6
	SBWANT IS UPPER LT					32 SWITCH POS CHANGES											
50 24 00	TDRS -171	234.1	-14.1	159.8	178.6	87.2	178.5	156.3	16.1	323.4	86.4	345.4	179.9	2.8		339.2	352.1
	SBWANT IS LOWER LT					33 SWITCH POS CHANGES											

Table XIII (Continued)

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET			EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
HRS	MIN	SECS		ALT N MI	LAT DEG	LONG DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG	ROLL DEG	
50	25	00	TDRS -171	234.3	-11.0	-157.8	179.4	83.3	299.3	156.3	16.1	83.4	282.7	74.2	185.8	356.7	17.0	346.3	
			SBWANT IS LOWER RT				34 SWITCH POS CHANGES												
50	25	30.0	TDRS -171	232.0	-9.4	-156.7	179.5	81.4	359.5	156.3	16.1	143.4	273.6	14.6	180.1	351.4	20.4	7.0	
			SBWANT IS UPPER RT				35 SWITCH POS CHANGES												
50	26	30.0	TDRS -171	232.1	-6.2	-154.7	179.7	77.5	119.6	156.3	16.1	263.4	77.3	285.8	169.1	6.1	343.9	13.8	
			SBWANT IS UPPER LT				36 SWITCH POS CHANGES												
50	27	00	TDRS -171	232.1	-4.6	-153.7	179.7	75.6	179.7	156.3	16.1	323.4	86.4	345.4	179.9	14.4	340.0	353.8	
			SBWANT IS LOWER LT				37 SWITCH POS CHANGES												
50	28	00	TDRS -171	232.2	-1.4	-151.8	179.8	71.7	299.7	156.3	16.1	83.4	282.7	74.2	196.0	351.1	15.1	346.1	
			SBWANT IS LOWER RT				38 SWITCH POS CHANGES												
50	28	30.0	TDRS -171	232.3	.2	-150.8	179.8	69.8	359.7	156.3	16.1	143.4	273.6	14.6	180.1	339.8	19.5	5.4	
			SBWANT IS UPPER RT				39 SWITCH POS CHANGES												
50	29	30.0	TDRS -171	232.4	3.4	-148.8	179.8	65.9	119.8	156.3	16.1	263.4	77.3	285.8	158.8	11.7	345.8	13.9	
			SBWANT IS UPPER LT				40 SWITCH POS CHANGES												
50	30	00	TDRS -171	232.5	5.0	-147.9	179.8	64.0	179.8	156.3	16.1	323.4	86.4	345.4	179.9	26.0	341.0	356.3	
			SBWANT IS LOWER LT				41 SWITCH POS CHANGES												
50	31	00	TDRS -171	232.7	6.1	-145.9	179.9	60.1	299.8	156.3	16.1	83.4	282.7	74.2	206.5	345.7	13.3	346.1	
			SBWANT IS LOWER RT				42 SWITCH POS CHANGES												
50	31	30.0	TDRS -171	232.0	9.7	-144.8	179.9	58.2	359.8	156.3	16.1	143.4	273.5	14.6	180.1	328.2	18.5	3.9	
			SBWANT IS UPPER RT				43 SWITCH POS CHANGES												
50	32	30.0	TDRS -171	232.2	12.9	-142.8	179.9	54.3	119.8	156.3	16.1	263.4	77.3	285.8	148.1	16.9	347.6	13.8	
			SBWANT IS UPPER LT				44 SWITCH POS CHANGES												
50	33	00	TDRS -171	232.3	14.5	-141.8	179.9	52.4	179.8	156.3	16.1	323.4	86.5	345.4	179.9	37.6	342.0	356.7	
			SBWANT IS LOWER LT				45 SWITCH POS CHANGES												
50	34	00	TDRS -171	232.4	17.6	-139.6	179.9	48.5	299.9	156.3	16.1	83.4	282.7	74.2	217.5	340.7	14.6	346.3	
			SBWANT IS LOWER RT				46 SWITCH POS CHANGES												
50	34	30.0	TDRS -171	232.5	19.1	-138.5	179.9	46.6	359.9	156.3	16.1	143.4	273.5	14.6	180.1	316.6	17.4	2.7	
			SBWANT IS UPPER RT				47 SWITCH POS CHANGES												
50	35	30.0	TDRS -171	232.6	22.2	-136.2	179.9	42.7	119.9	156.3	16.1	263.4	77.3	285.8	136.8	21.5	349.2	13.5	
			SBWANT IS UPPER LT				48 SWITCH POS CHANGES												
50	36	00	TDRS -171	232.7	25.7	-135.1	179.9	40.8	179.9	156.3	16.1	323.4	86.5	345.4	179.9	49.2	343.1	357.8	
			SBWANT IS LOWER LT				49 SWITCH POS CHANGES												

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

Table XIII. (Continued)

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET HRS MN SECS	EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
		ALT N MI	LAT DEG	LOX DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG		
50 37 00	TDRS -171 SBWANT IS LOWER RT	232.8	26.7	-132.6	179.9	36.9	299.9	156.3	16.1	83.4	282.7	74.2	229.1	336.5	10.1	346.8	
50 37 30.0	TDRS -171 SBWANT IS UPPER RT	233.8	28.2	-131.3	179.9	35.0	359.9	156.3	16.1	143.4	273.5	14.6	180.2	305.0	16.3	1.8	
50 38 30.0	TDRS -171 SBWANT IS UPPER LT	233.7	31.1	-128.7	179.9	31.1	119.9	156.3	16.1	263.4	77.3	285.8	124.9	25.3	350.5	12.9	
50 39 00	TDRS -171 SBWANT IS LOWER LT	233.6	32.6	-127.3	179.9	29.2	179.9	156.3	16.1	323.4	86.5	345.4	179.8	60.8	344.3	358.5	
50 40 00	TDRS -171 SBWANT IS LOWER RT	233.5	35.4	-124.3	179.9	25.4	299.9	156.3	16.1	83.4	282.7	74.2	241.3	333.2	9.0	347.4	
50 40 30.0	TDRS -171 SBWANT IS UPPER RT	233.4	38.7	-122.8	179.9	23.4	359.9	156.3	16.1	143.4	273.5	14.6	180.2	293.4	15.2	1.3	
50 41 30.0	TDRS -171 SBWANT IS UPPER LT	233.3	39.4	-119.5	179.9	19.6	119.9	156.3	16.1	263.4	77.3	285.8	112.3	28.0	351.4	12.2	
50 42 00	TDRS -171 SBWANT IS LOWER LT	233.2	40.7	-117.7	179.9	17.6	179.9	156.3	16.1	323.4	86.5	345.4	179.8	72.4	345.4	358.8	
50 43 00	TDRS -171 SBWANT IS LOWER RT	232.9	43.2	-114.0	179.9	13.8	299.9	156.3	16.1	83.4	282.7	74.2	254.2	331.0	8.2	348.3	
50 43 30.0	TDRS -171 SBWANT IS UPPER RT	236.3	44.3	-112.0	179.9	11.8	359.9	156.3	16.1	143.4	273.5	14.6	180.3	281.8	14.1	1.1	
50 44 30.0	TDRS -171 SBWANT IS UPPER LT	236.4	46.6	-107.8	179.9	8.0	119.9	156.3	16.1	263.4	77.3	285.8	99.2	29.6	352.0	11.3	
50 45 00	TDRS -41 SBWANT IS LOWER RT	236.4	47.6	-105.5	179.9	6.0	179.9	156.3	16.1	323.4	86.5	345.4	179.5	84.0	127.6	348.5	
50 45 30.0	TDRS -41 SBWANT IS UPPER RT	236.4	48.6	-103.2	179.9	4.1	239.9	156.3	16.1	23.4	85.1	45.3	265.3	30.0	136.8	35.1	
50 46 00	TDRS -41 SBWANT IS UPPER LT	236.4	49.6	-100.7	179.9	2.2	300.0	156.3	16.1	83.4	282.7	74.2	267.5	330.1	200.2	50.7	
50 47 00	TDRS -41 SBWANT IS LOWER LT	236.4	51.3	-95.6	179.9	358.3	60.0	156.3	16.1	203.4	274.9	314.7	88.1	330.0	223.5	324.5	
50 47 30.0	TDRS -41 SBWANT IS LOWER RT	236.4	52.0	-92.8	179.9	356.4	120.0	156.3	16.1	263.4	77.3	285.8	85.8	29.9	159.2	309.0	
50 48 30.0	TDRS -41 SBWANT IS UPPER RT	236.2	53.3	-87.1	179.9	352.5	240.0	156.3	16.1	23.4	85.1	45.3	278.6	29.7	136.1	35.8	

Table XIII. (Continued)

SHUTTLE ATTITUDE AND POINTING TIME LINE

GET			EVENT	VEHICLE POSITION			LVLH ATTITUDE			ECI ATTITUDE			LOOK ANGLES TO SUN		LOOK ANGLES TO EARTH		LOOK ANGLES TO TARGET		
HRS	MM	SECS		ALT N MI	LAT DEG	LON DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	ROLL DEG	YAW DEG	PITCH DEG	YAW DEG	PITCH DEG	TARG ID	YAW DEG	PITCH DEG
50	49	00	TDRS -41	236.2	53.9	-84.1	179.9	350.6	300.0	156.3	16.1	83.4	282.7	74.2	280.8	330.5		201.2	51.4
			SBWANT IS UPPER LT				67 SWITCH POS CHANGES												
50	50	00	TDRS -41	237.8	54.7	-77.9	179.9	346.7	60.0	156.3	16.1	203.4	274.9	314.7	74.7	330.9		224.4	324.0
			SBWANT IS LOWER LT				68 SWITCH POS CHANGES												
50	50	30.0	TDRS -41	237.7	54.9	-74.7	179.9	344.8	120.0	156.3	16.1	263.4	77.3	285.8	72.5	28.8		158.5	308.2
			SBWANT IS LOWER RT				69 SWITCH POS CHANGES												
50	51	30.0	TDRS -41	237.6	55.2	-68.3	179.9	340.9	240.0	156.3	16.1	23.4	85.1	45.3	291.8	28.2		135.0	36.2
			SBWANT IS UPPER RT				70 SWITCH POS CHANGES												
50	52	00	TDRS -41	237.5	55.2	-65.0	179.9	339.0	300.0	156.3	16.1	83.4	282.7	74.2	293.9	332.2		201.6	52.3
			SBWANT IS UPPER LT				71 SWITCH POS CHANGES												
50	53	00	TDRS -41	237.2	54.9	-58.6	179.8	335.1	60.0	156.3	16.1	203.4	274.9	314.7	61.8	333.0		225.6	323.8
			SBWANT IS LOWER LT				72 SWITCH POS CHANGES												
50	53	30.0	TDRS -41	237.1	54.6	-55.4	179.8	333.2	120.0	156.3	16.1	263.4	77.3	285.8	59.7	26.5		158.4	307.2
			SBWANT IS LOWER RT				73 SWITCH POS CHANGES												
50	54	30.0	TDRS -41	236.8	53.8	-49.2	179.8	329.3	240.0	156.3	16.1	23.4	85.1	45.3	304.4	25.4		133.8	36.2
			SBWANT IS UPPER RT				74 SWITCH POS CHANGES												
50	55	00	TDRS -41	236.6	53.3	-46.2	179.8	327.4	300.0	156.3	16.1	83.4	282.7	74.2	306.5	335.1		201.5	53.3
			SBWANT IS UPPER LT				75 SWITCH POS CHANGES												
50	56	00	TDRS -41	237.0	52.8	-40.5	179.8	323.5	60.0	156.3	16.1	203.4	274.9	314.7	49.5	336.3		226.9	323.9
			SBWANT IS LOWER LT				76 SWITCH POS CHANGES												
50	56	30.0	TDRS -41	236.8	51.2	-37.8	179.8	321.6	120.0	156.3	16.1	263.4	77.3	285.8	47.5	23.1		158.8	306.1
			SBWANT IS LOWER RT				77 SWITCH POS CHANGES												
50	57	30.0	TDRS -41	236.3	49.5	-32.6	179.8	317.7	240.0	156.3	16.1	23.4	85.1	45.3	316.4	21.6		132.4	35.9
			SBWANT IS UPPER RT				78 SWITCH POS CHANGES												
50	58	00	TDRS -41	236.1	48.5	-30.2	179.8	315.8	300.0	156.3	16.1	83.4	282.7	74.2	318.4	338.9		200.7	54.4
			SBWANT IS UPPER LI				79 SWITCH POS CHANGES												
50	59	00	TDRS -41	235.7	46.5	-25.6	179.7	311.9	60.0	156.3	16.1	203.4	274.9	314.7	37.9	340.6		228.3	324.4
			SBWANT IS LOWER LT				80 SWITCH POS CHANGES												
50	59	30.0	TDRS -41	235.5	45.4	-23.5	179.7	310.0	120.0	156.3	16.1	263.4	77.3	285.8	35.9	18.8		159.9	305.1
			SBWANT IS LOWER RT				81 SWITCH POS CHANGES												

Table XIII. (Continued)

Table IV shows an ECI pitch of 0.5° per second over a one orbit period. In this case only 16 switch changes occur. The ECI yaw and roll angles appear to vary during this simulation. The yaw-pitch-roll sequence results in apparent changes in the yaw and roll angles; yet, when taken in that sequence the resulting attitude maneuver is an ECI pitch of 0.5° per second. Similar results are obtained in Table V with 17 switch position changes for an ECI yaw of 0.5° per second. Table VI shows 32 position changes for thermal conditioning with the X-axis of the orbiter perpendicular to the Orbiter-sun line and a barbeque rate of 0.5° per second. Tables VIII - IX show 2-4 switch changes for star tracking over a 60 minute period. Target 522 is designated as M82, target 522 is designated as M42 (Oeion) and target 530 is designated as W51 (W3). The ECI inertial hold of Table X resulted in 2 switch position changes and the heads down LVLH earth pointing results in 3 changes in Table XI. The LVLH simulation of Table XII resulted in 6 position changes for at 100 minutes period. And finally in Table XIII the worst case orbital angular roll rate of 2° per second is seen to result in 81 changes for a 60 minute period. A summary is given in Table XIV to show switch changes which were encountered for each activity simulated. Also included in Tables XIV are the required switch update periods which correspond to a 5° worst case overlap condition in which the Orbiter could maneuver 5° out of the limits of its preferred pattern into the optimum region of an adjacent antenna. It is noted that the actual

PARAMETER ORBITER ACTIVITY	ANGULAR RATE	ATTITUDE CONSTRAINTS	AVERAGE TIME BETWEEN CHANGES	AVERAGE CHANGES PFR HOUR	REQUIRED UPDATE PERIOD*
INERTIAL HOLD	0	ECI fixed	15-45 min.	2.65	50-150 sec
LVLH	0	$\theta = 0^\circ$ $\psi = 0^\circ$ $\phi = 0^\circ$	16.7 min.	3.6	56 sec
STAR TRACKING	0	ECI fixed	15-30 min.	2.67	50-100 sec.
EARTH POINTING	0	LVLH HEADS DOWN	30 min.	2	100 sec.
ROLL MANEUVER	0.5° per sec	ECI $\theta = \text{const.}$ $\psi = \text{const.}$	2.9 min.	20.7	9.7 sec.
ROLL MANEUVER	2° per sec	ECI $\theta = \text{const.}$ $\psi = \text{const.}$	44.44 sec.	81	2.47 sec.
YAW MANEUVER	0.5° per sec	ECI $\theta = \text{const.}$ $\phi = \text{const.}$	5.3 min.	11.3	17.6 sec.
PITCH MANEUVER	0.5° per sec	ECI $\psi = \text{const.}$ $\phi = \text{const.}$	5.6 min.	10.7	18.75 sec.
THERMAL CONDX	0.5° per sec	X-axis to sunline	2.8 min.	21.4	9.4 sec.
** LANDING	5° per sec bank	N/A	10 sec.*	N/A	1 sec.*
** LAUNCH	9° per sec roll	N/A	15-22.5 sec.*	N/A	.55 sec.*

*5° OVERLAP

** ANALYTICAL RESULT-WORST CASE

TABLE XIV. SUMMARY OF S-BAND QUAD ANTENNA SWITCHING REQUIREMENTS

update period used will limit the amount of switch "chatter" which could occur if the look angle happens to oscillate between two optimum regions, so in effect it controls the deadband of the control system.

VI. CONCLUSIONS

Mission 2 activities are discussed in Section III with each orbital activity categorized and expressed as a percent of the total mission time excluding launch and landing. The percentages in Table I do not account for 100% of the mission as the remaining 2.5 to 6 percent is not defined at this time. To assess the total number of switch position changes for a mission the formula below is used.

$$N_T = N_e + \sum_{a=1}^N N_a P_a T \quad (1)$$

Where T is the total mission time in hours excluding launch and landing which is 161 hours 20 minutes and 3 seconds.

N_a is the number of switch position changes per hour for a particular activity (a = 1 to N)

P_a is the probability (percent total time x 100) for a particular activity (a = 1 to N)

N_e is an estimate of switch position changes during launch, landing and times not accounted for.

The estimate N_e is based on 4 switch changes per hour during unaccounted for times and 34 changes during launch and landing the estimate for the number of switch position changes N_e is 50 for Option 1, 74 for Option 2 and 60 for Option 3. A listing of the total switch changes for Mission 2, Options 1, 2 and 3 is given in Table XV using Equation (1). The average number of switch

	TOTAL NUMBER OF SWITCH CHANGES	AVERAGE TIME BETWEEN CHANGES	MIN. IN ORBIT UPDATE PERIOD*	AVE UPDATE PERIOD*
OPTION 1	1466	6.8 min	2.4 sec**	22.6 sec
OPTION 2	1247	7.97 min	2.4 sec**	26.6 sec
OPTION 3	1221	8.13 min	2.4 sec**	27.1 sec

*5° OVERLAP

** BASED ON 2° PER SECOND ANGULAR RATE IN ORBIT

TABLE XV. TOTAL SWITCH CHANGES FOR MISSION 2

position changes for Mission 2 is seen to be approximately 1300. Also, the average time between changes and the minimum update period is given.

An examination of the data in Section V shows that the maximum orbital angular roll rate of 2° per second will require an update period of approximately 2.4 seconds without exceeding the 5° overlap limit. For the worst case conditions of launch and landing the required update period is reduced to 0.55 seconds and 1 second respectively. With an update of 2.4 seconds the associated Orbiter computer will be required to make a line-of-sight look angle calculation, a TDRS selection and an antenna selection approximately 248,000 times during Mission 2. For an update period of 1 second (worst case landing) 596,000 computer calculations would be required and for a 0.55 second update period (worst case launch) 1,083,000 calculations would be required. Since launch and landing are rather precisely defined, it is probably unreasonable to specify an update based on the worst case conditions of launch and landing. The use of a preprogrammed switching sequence or ground commands for antenna control are recommended for this period.

Another factor to be considered is that a 2° per second roll accounts for only a very small part of the mission and the required update period during the great majority of the activities is in the range of 30-50 seconds. Thus, the possibility of a variable update period should be considered. The shorter update period of 2.4 seconds could

be used when angular rates exceed a certain value. The variable update period would require the additional Orbiter computer input of angular rate.

After the development and use of an extensive computer program tool to simulate automatic switching of the S-Band Quad Antennas, the following conclusions may be reached. A 2.4 second update period should be used for S-Band Quad Antenna switching assuming a fixed update rate. If a variable update period is found to be practical, then the update period would be approximately 40 seconds during most activities with a change to 2.4 seconds during activities involving larger angular rates.

Future efforts involving the S-Band Quad Antennas should logically involve a more detailed study of the associated Orbiter computer interfaces and the further analysis for simulation of Missions 1, 3A and 3B along with a feasibility study for implementing a variable updated period.

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APPENDIX I

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1*      COMMON/ANGLES/ YAWTAR,PTCTAR,SBQANT(2),NCOUNT
2*      COMMON /ALONG/ SCLON,VARR2,WRTFLG
3*      CD1      **** PROGRAM SAPT ****
4*      CD1 THE SHUTTLE ATTITUDE AND POINTING TIMELINE PROCESSOR (SAPT) IS THE
5*      CD1 BASIC ATTITUDE AND POINTING TIMELINE GENERATION PROGRAM FOR
6*      CD1 SHUTTLE. SAPT CAN GENERATE ATTITUDE AND POINTING TIMELINE TABLES
7*      CD1 FOR DOCUMENTATION AND AT THE SAME TIME ALSO GENERATE ATTITUDE
8*      CD1 TAPES FOR SUBSYSTEM EVALUATIONS. THE BASIC INPUT IS IN THE FORM
9*      CD1 OF ATTITUDE AND POINTING DESCRIPTOR CARDS DEFINING THE NAME OF
10*     CD1 THE EVENT, THE EVENT TIME, AND DESCRIPTORS DEFINING THE ATTITUDE
11*     CD1 AND POINTING REQUIREMENTS FOR THE EVENT. THE PROGRAM REQUIRES
12*     CD1 A TRAJECTORY TAPE MOUNTED ON UNIT F, AN EPHEMERIS TAPE MOUNTED ON
13*     CD1 UNIT K, AND A PCF TAPE INCLUDING SAPT. IF AN ATTITUDE TAPE IS ALSO
14*     CD1 TO BE GENERATED, A SAVE TAPE MUST BE MOUNTED ON THE TAPE UNIT
15*     CD1 DEFINED BY ITAPE. THE FIRST INPUT CARD AFTER THE XOT CARD MUST BE
16*     CD1 A BASE CARD GIVING THE BASE DATE TO MEASURE G.E.T. FROM AND THE
17*     CD1 TAPE UNIT AND PRINT INTERVAL IF A ATTITUDE TAPE IS TO BE GENERATED
18*     CD1 THERE IS ALSO AN INPUT FLAG WHICH DEFINES A TAPE UNIT TO WRITE A
19*     CD1 DESCRIPTOR TIMELINE. AFTER THE BASE CARD, THE ATTITUDE AND POINT-
20*     CD1 ING DESCRIPTOR CARDS FOLLOW. THERE MAY BE UP TO 4 CARDS REQUIRED
21*     CD1 TO DEFINE EACH EVENT. THE FIRST CARD IN EACH EVENT SET IS THE
22*     CD1 EVENT NAME CARD. UP TO 4 LINES OF EVENT DESCRIPTION CAN BE DEFINED
23*     CD1 WITH 18 CHARACTERS TO EACH LINE. THE NEXT CARD IS THE ATTITUDE
24*     CD1 REQUIREMENTS CARD GIVING THE EVENT TIME AND ATTITUDE ONLY REQUIR-
25*     CD1 MENTS. IF THERE ARE NO POINTING REQUIREMENTS, THIS IS THE LAST
26*     CD1 CARD OF THE SET. IF THERE ARE POINTING REQUIREMENTS, 1 OR 2 MORE
27*     CD1 CARDS ARE REQUIRED DEPENDING ON THE NUMBER OF POINTING REQUIR-
28*     CD1 MENTS. EVENT SETS SHOULD BE STACKED IN CHRONOLOGICAL ORDER. THE
29*     CD1 LAST CARD OF THE DECK IS LABELED *END CARD, STARTING IN CC 1.
30*     CD1 THE INPUT PARAMETER DEFINITIONS FOR THE BASE CARD ARE GIVEN BELOW.
31*     CD1 THE INPUT DEFINITIONS FOR THE EVENT SETS ARE GIVEN IN SUBROUTINE

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*NEW
*NEW

REPRODUCIBILITY OF THE
ORIGINAL PAGE IS POOR

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32* CD1 'CARDIN'
33* CD2 BASE CARD DEFINITIONS -
34* CD2 NBASYR,NBASMT,NBASDY,NBASHR,NBASMN,BASSEC
35* CD2 - BASE G.M.T. THAT G.E.T. IS MEASURED FROM (NOMINALLY THE
36* CD2 LAUNCH G.M.T.
37* CD2 ITAPE - UNIT ON WHICH THE COMMON FORMAT ATTITUDE TAPE IS TO BE
38* CD2 WRITTEN. IF ZERO, NO TAPE IS REQUIRED.
39* CD2 GMTINT - TIME INCREMENT BETWEEN PRINT POINTS DESIRED FOR THE
40* CD2 COMMON FORMAT TAPE. (MINUTES)
41* CD2 IUREAD - UNIT ON WHICH THE INPUT ATTITUDE DESCRIPTOR FILE WILL BE
42* CD2 MOUNTED. IF ZERO, ALL INPUT WILL BE MANUAL.
43* CD2 IUWRT - UNIT ON WHICH THE REVISED ATTITUDE DESCRIPTOR FILE IS TO
44* CD2 BE WRITTEN. IF ZERO, NO TAPE IS TO BE WRITTEN.
45* CD2 IHRST,MINST,SECT
46* CD2 - GET TO START ATTITUDE TIMELINE IF A DESCRIPTOR FILE HAS
47* CD2 BEEN PROVIDED. IF NOT INPUT, THE START OF THE DESCRIPTOR
48* CD2 FILE, OR THE FIRST MANUAL INPUT, WHICHEVER IS EARLIER,
49* CD2 WILL BE USED.
50* CD2 IHREND,MINEND,SECEND
51* CD2 - GET TO END ATTITUDE TIMELINE IF A DESCRIPTOR FILE HAS
52* CD2 BEEN PROVIDED. IF NOT INPUT, THE END OF THE DESCRIPTOR
53* CD2 FILE, OR THE LAST MANUAL INPUT, WHICHEVER IS LATER,
54* CD2 WILL BE USED.
55* C
56* INTEGER AITSYS,ALIGN
57* REAL IANG1,IANG2,IANG12,IANG22
58* DOUBLE PRECISION BASEJD,BASTIM,IGFT,GHT,GMTP
59* DOUBLE PRECISION GMTNXT,TSMCAL,TNEXT,TIMTAR
60* DOUBLE PRECISION XIMUDP,YIMUDP,ZIMUDP,RKMDP(3),VKMOP(3),DV(3)
61* DOUBLE PRECISION GET
62* DOUBLE PRECISION RSUNDP(3)
63* REAL LIFTKG,ISPSEC
64* REAL KX,KY,KZ,NUZ,MINTIM
65* DIMENSION DUMHAT(15),R(3),V(3),RSUN(3),VSUN(3),RMOON(3),VMOON(3)
66* DIMENSION TARGET(3),TDVFC(3),ATTLV(3,3),ATTI(3,3),SUNVEC(3),
67* * PB(3),COEVEC(3),ATTECI(3,3)
68* DIMENSION RKM(3),VKM(3),RKMU(3),RSUNU(3)
69* DIMENSION ATTS(3,3)
70* DIMENSION SPARE(14),RSUNKH(3)
71* COMMON/BASDT/ BASEJD,BASTIM,NBASR,NBASMT,NBASDY,NBASHR,NBASMN,
72* * BASSEC
73* COMMON/ATPCOM/ GHTP,ATTP(3,3)
74* COMMON/JERRCM/ JERR
75* COMMON / VHLB / KX,KY,KZ,NUZ,BETA,RASUN,DFCSUN,RMOON,DECMON
76* COMMON/SHCAL/ TSMCAL,RSUN,VSUN,RMOON,VMOON
77* COMMON/CARDS/ IUR,MN,SECS,ATTSYS,IMUNUM,IVMS,ANG1,ANG2,ANG3
78* * IHOLD,XRATE,YRATE,ZRATE,IPTFLG,ALIGN,INSTID,IANG1,IANG2
79* * ITYPE1,IDEF1,VAR1,VAR2,VAR3,ITARG1,VAR4,IDIN2,IANG12,IANG22
80* * ITYPE2,IDEF2,VAR12,VAR22,VAR32,ITARG2,VAR42
81* COMMON/IVCOM/ ITARG,ALPHAT,BETAT,WVEC(3),ALPHAC,BFTAC,AT,
82* * TLOM,TLAT,ITYPE,IDEF
83* COMMON/EVNCOM/NEVENT(12),NAHTAR
84* COMMON / REVST / REV
85* C REV - REV NUMBER OF VEHICLE FROM START OF MISSION
86* COMMON /TRJCNV / A,E,DINC,ARGPER,ASCNOD,AM,DMEAN,TRUEAN,PERIOD
87* DATA RTD/57.2957795/,SECPHR/3600,n/
88* DATA DUMHAT / 15.0 0.0 /,MINTIM / -1.0 /
89* DATA ITAPE / 15 /

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SAPT PROGRAM (CONTINUED)

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90*   DATA NBLANK/6H /,MAXLIN/56/,NUMLIN/56/
91*   DATA NCARD / 6H /
92*   C      EGRAVC = GRAVITATIONAL CONSTANT OF THE EARTH ( KM.03 / SEC.00 )
93*   DATA EGRAVC / 98.69119999 /
94*   C      DV = SENSED VELOCITY CHANGE OVER TIME INTERVAL BETWEEN PRESENT
95*   C      AND PREVIOUS RECORD
96*   DATA DV / 0.00000000000000 /
97*   DATA SPARE / 14 * 0.0 /
98*   DATA NUMENT/1/
99*   ORR = 1
100*  WRTFLG = 0.0
101*  SBOANT(1) = 0.0
102*  PRTFLG = 0.0
103*  LIFTKG = 0.0
104*  DYNPR = 0.0
105*  DRAGKG = 0.0
106*  WGTKG = 0.0
107*  THRKG = 0.0
108*  ISPSEC = 0.0
109*  CALL INTMSG(0,0,0)
110*  C      BASE CARD VALUFS AND FORMAT FOLLOW
111*  C
112*  READ(5,910),NBASYSR,NBASMT,NBASDY,NBASHR,NBASMN,BASCSEC,
113*  * ITAPE,GMTINT,IUREAD,IUWRT,IHRST,MINST,SECST,IHREND,MINEND,SPCEND
114*  *,IASTP
115*  910 FORMAT(1X,14,4(1X,12),1X,F5.2,1X,12,1X,F6.0,2(1X,12),
116*  * 2(1X,14,1X,12,1X,F5.2),12X,11)
117*  C
118*  WRITE(6,900)
119*  900 FORMAT(1H1)
120*  CALL DATEC
121*  GETST = IHRST + MINST/60.0 + SECST/3600.0
122*  GETEND = IHREND + MINEND/60.0 + SECEND/3600.0
123*  IF(GETEND .LE. GETST) GETEND = 200.0
124*  CALL INPROC(IUREAD,GETST,GETEND,NUMENT)
125*  100 CONTINUE
126*  C      THE CALL TO GETDEC OBTAINS THE FIRST EVENT SET
127*  C
128*  CALL GETDES(IUREAD,ISTOP)
129*  C
130*  CALL CROFIL(I)
131*  IF(ISTOP .GT. 0) GO TO 350
132*  IF(IERR .GT. 0) STOP
133*  120 CONTINUE
134*  JPCARD = 1
135*  TGET = IHR + MIN/60 + SECS/3600
136*  GMT = TGET + BASTIM
137*  150 CONTINUE
138*  CALL TIMEC(GMT,1DAYN,IHRN,MINN,SECN)
139*  TIMTAP = GMT - BASTIM
140*  CALL TIMEC(TIMTAP,1DT,1HT,1MT,1ST)
141*  1HT = 1DT*24 + 1HT
142*  ITAPE = SNGJ(TIMTAP)
143*  IF(IHRN .GT. 2) .AND. (1DAYN - 1DCALL) .EQ. 1) 1DCALL = 1DCALL-1
144*  IF((1DAYN - 1DCALL) .GE. 9 .AND. (1DAYN - 1DCALL) .LT.2) GO TO 200
145*  1DCALL = 1DAYN
146*  IF(IHRN .LT. 2) 1DCALL = 1DAYN - 1
147*  CALL INITAL(1DCALL,ISTOP)

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*NEW
*NEW
*NEW

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***1

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ORIGINAL PAGE IS POOR

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148*      IF(IISTOP .GT. 0) STOP
149*      200 CONTINUE
150*      IF(ALIGN .NF. 1) GO TO 220
151*      CALL CRDFIL(2)
152*      CALL ATTDEF(GMT,ATTP)
153*      GMTP = GMT
154*      CALL CRDFIL(1)
155*      220 CONTINUE
156*      IF (ATTSYS .EQ. NBLANK) IBLANK = 1
157*      IF(ATTSYS .FO. NBLANK) CALL CRDFIL(2)
158*      CALL ATTDEF(GMT,ATTECI)
159*      CALL CRDFIL(1)
160*      IF(ATTSYS .NE. NBLANK) GO TO 225
161*      DO 223 I=1,3
162*      DO 223 J=1,3
163*      ATTI(I,J) = ATTECI(I,J)
164*      223 CONTINUE
165*      CALL EULER(ATTI,IVMS,ANG1,ANG2,ANG3,ATTECI)
166*      225 CONTINUE
167*      CALL NADIR(AMT,SCLON,GDLAT,ALT)
168*      IF(IPTFLG .FO. 0) GO TO 250
169*      CALL FILTYC(1)
170*      IF(ATTSYS .EQ. NBLANK) CALL CRDFIL(2)
171*      CALL TARVEC(GMT,NEWVEL,TARGET,TDVEC)
172*      CALL BPCR(1,ATTECI,TARGET,PB)
173*      CALL YWPTCH(2,VAHTAR,PTCTAR,PB)
174*      250 CONTINUE
175*      ITYPE = 1
176*      IDEF = 0
177*      CALL TARVEC(GMT,NEWVEL,SUNVEC,TDVEC)
178*      CALL BPCR(1,ATTECI,SUNVEC,PB)
179*      CALL YWPTCH(2,VAWSUN,PTCSUN,PB)
180*      ITYPE = 1
181*      IDEF = 1
182*      CALL TARVEC(GMT,NEWVEL,COEVEC,TDVEC)
183*      CALL BPCR(1,ATTECI,COEVEC,PB)
184*      CALL YWPTCH(2,VAWCOE,PTCCOE,PB)
185*      CALL ACP(1,2,2,4,1,3,3,IMUNUM,IMUNUM,GMT,GMT,ATTECI,TDVEC)
186*      • OMG,0,0,0,0,0,YRATE,7RATE,XRATE,ATTLV,YOFF,ZOFF,XOFF)
187*      ZLVLH = ZOFF*RTD
188*      YLVLH = YOFF*RTD
189*      XLVLH = XOFF*RTD
190*      IF(ZLVLH .LT. -0.05) ZLVLH = 360.0 + ZLVLH
191*      IF(YLVLH .LT. -0.05) YLVLH = 360.0 + YLVLH
192*      IF(XLVLH .LT. -0.05) XLVLH = 360.0 + XLVLH
193*      IF(IHOLD .NF. 1) GO TO 260
194*      CALL ZLVDDT(7,YLVLH,YLVLH,ZLVLH,XRATE,YRATE,ZRATE,
195*      XRATE = XRATE*/SECPHR/RTD)
196*      YRATE = YRATE*/SECPHR/RTD)
197*      ZRATE = ZRATE*/SECPHR/RTD)
198*      260 CONTINUE
199*      IF(IMUNUM .FO. 0) IMUNUM=1
200*      M=3
201*      IF(IASTP .EQ. 1) M=2
202*      CALL ACP(1,2,2,4,3,3,M,IMUNUM,IMUNUM,GMT,GMT,ATTECI,TDVEC)
203*      • OMG,0,0,0,0,0,YRATE,7RATE,XRATE,ATTLV,YOFF,ZOFF,XOFF)
204*      ZIMU = ZOFF*RTD
205*      YIMU = YOFF*RTD

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SAPT PROGRAM (CONTINUED)


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@      FOR,G  SAPT,SAPT
!
206*      XIMU = XOFF*RTD
207*      IF(ZIMU .LT. -0.05) ZIMU = 360.0 + ZIMU
208*      IF(YIMU .LT. -0.05) YIMU = 360.0 + YIMU
209*      IF(XIMU .LT. -0.05) XIMU = 360.0 + XIMU
210*      YAWTAR = YAWTAR*RTD
211*      PTCTAR = PTCTAR*RTD
212*      YAWSUN = YAWSUN*RTD
213*      PTCSUN = PTCSUN*RTD
214*      YAWCOE = YAWCOE*RTD
215*      PTCCOE = PTCCOE*RTD
216*      IF(SCLON .GT. 180.0) SCLON = SCLON - 360.0
217*      NUMCOM = 1
218*      DO 280 I=2,4
219*      DO 280 J=1,3
220*      IJ = (I-1)*3. + J
221*      IF(NEVENT(IJ) .NE. NBLANK) NUMCOM = I
222*      280 CONTINUE
223*      CALL TRAJCT (GMT,R,V,JERR)
224*      IF (JERR .GT. 0) STOP
225*      RMAG = VECMG(R)
226*      IF (ITAPE .EQ. 0) GO TO 288
227*      RECFLG = 1.0
228*      C      RECFLG- LAST RECORD IN FILE FLAG. IF LESS THAN 0.0 THIS IMPLIES
229*      C      LAST RECORD IN FILE
230*      C      PRTEFLG- FLAG INDICATING TYPE OF PRINT POINT
231*      CALL GMTCAL (GMT,NBASR,YEAR,GMONTH,GDAY,GHR,GMIN,GSEC)
232*      C      SUBROUTINE GMTCAL WILL COMPUTE GREENWICH MEAN TIME IN YEAR,
233*      C      MONTH, DAY, HOUR, MINUTES, AND SECONDS FROM THE BEGINNING OF THE
234*      C      BASE YEAR (NBASR).
235*      GET = GMT - BASRIM
236*      C      GET = TIME IN HOURS FROM LAUNCH
237*      DO 281 J=1,3
238*      RKM(J) = R(J) * 6378.16
239*      C      RKM = VEHICLE RADIUS VECTOR (ECI) IN KM
240*      VKM(J) = V(J) * 16378.16 / 3600.0
241*      C      VKM = VEHICLE VELOCITY VECTOR (ECI) IN KM/SEC
242*      RSUNDP(J) = DBLE(RSUN(J)) * 6378.16000
243*      RSUNKM(J) = SNGL(RSUNDP(J))
244*      C      RSUNKM = SUN VECTOR IN KM (ECI)
245*      281 CONTINUE
246*      DO 285 I=1,3
247*      RKMDP(I) = DBLE(RKM(I))
248*      C      RKMDP = VEHICLE RADIUS VECTOR (DOUBLE PRECISION)
249*      VKMDP(I) = DBLE(VKM(I))
250*      C      VKMDP = VEHICLE VELOCITY VECTOR (DOUBLE PRECISION)
251*      285 CONTINUE
252*      XIMUDP = DBLE(XIMU) / 57.295779500
253*      YIMUDP = DBLE(YIMU) / 57.295779500
254*      ZIMUDP = DBLE(ZIMU) / 57.295779500
255*      C      YIMUDP,ZIMUDP,XIMUDP= PITCH, YAW, AND ROLL GIMBAL ANGLES (RADIANS)
256*      C      WITH RESPECT TO INPUT REFMMAT
257*      RMAGKM = VECMG(RKM)
258*      C      RMAGKM = VEHICLE GEOCENTRIC RADIUS (KM)
259*      CALL UNVEC (RKM,RKMU)
260*      C      RKMU = VEHICLE UNIT RADIUS VECTOR (ECI)
261*      DECGCN = ASIN (RKMU(3))
262*      C      DECGCN = VEHICLE GEOCENTRIC DECLINATION (RADIANS)
263*      RASGCM = ATAN2 (RKMU(2),RKMU(1))

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SAPT PROGRAM (CONTINUED)

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264* C  RASGCM = VEHICLE GEOCENTRIC RIGHT ASCENSION (RADIAN$)
265* C  VMAGKM = VELOCITY (VKM)
266* C  VMAGKM = VEHICLE INERTIAL VELOCITY VECTOR MAGNITUDE (KM/SEC)
267* C  VALTKM = ALT * (6378.16 / 3443.9308855)
268* C  VALTKM = VEHICLE ALTITUDE ABOVE ORLATE EARTH (KM)
269* C  VEHGDL = GDIAT / RTD
270* C  VEHGDL = VEHICLE GEODETIC LATITUDE (RADIAN$)
271* C  VEHLON = SCION / RTD
272* C  VEHLON = VEHICLE LONGITUDE (RADIAN$)
273* C  CALL FPAZ (GMT,GAMI,PSII,VELREL,GAMREL,PSIREL,GRAS)
274* C  GAMI = INERTIAL FLIGHT PATH ANGLE (RADIAN$)
275* C  PSII = INERTIAL AZIMUTH (RADIAN$)
276* C  VELREL = RELATIVE VELOCITY VECTOR MAGNITUDE (KM/SEC)
277* C  GAMREL = RELATIVE FLIGHT PATH ANGLE (RADIAN$)
278* C  PSIREL = RELATIVE AZIMUTH (RADIAN$)
279* C  GRAS = RIGHT ASCENSION OF GREENWICH (RADIAN$)
280* C  AKM = A * (1.0 / 3280.833)
281* C  AKM = SEMIMAJOR AXIS (KM)
282* C  E = ECCENTRICITY
283* C  RADINC = DINC / RTD
284* C  RADINC = INCLINATION TO ORBITAL PLANE (RADIAN$)
285* C  ASCNR = ASCNOD / RTD
286* C  ASCNR = RIGHT ASCENSION OF ASCENDING NODE (RADIAN$)
287* C  ARGPR = ARGPER / RTD
288* C  ARGPR = ARGUMENT OF PERIGEE (RADIAN$)
289* C  TRUEAN = TRUE ANOMALY (RADIAN$)
290* C  PERIOD = ORBITAL PERIOD (SECONDS)
291* C  SHADOW = 1.0 VEHICLE IN DIRECT SUNLIGHT
292* C  SHADOW = -1.0 VEHICLE IN DARKNESS
293* C  SHADOW = 1.0
294* C  CALL UNVEC (RSIN,RSUNU)
295* C  COSANG = DOT (OKMU,RSUNU)
296* C  IF ( ABS(COSANG) .GE. 1.0) COSANG = SIGN(1.0,COSANG)
297* C  IF ( ACOS(COSANG) .GE. ( 180.0 - ASIN(1.0 / RHAG) ) ) SHADOW = -1.0
298* C  ATTECI = DIRECTION COSINES OF THE X,Y, AND Z BODY AXES WITH
299* C  RESPECT TO THE ECI COORDINATE SYSTEM
300* C  CALL ACP (1,2,4,5,3,3,IMUNUM,IMUNUM,GMT,GMT,ATTECI,TDVEC,
301* C  * OHG,0.0,D.0,0.0,0.0,0.0,0.0,0.0,ATTSR,ALSR,BTSR,GMSR)
302* C  ALSR = PITCH FROM SOLAR REFERENCE SYSTEM (ZYX EULER)
303* C  BTSR = YAW FROM SOLAR REFERENCE SYSTEM (ZYX EULER)
304* C  GMSR = ROLL FROM SOLAR REFERENCE SYSTEM (ZYX EULER)
305* C  YLVLHR = YLVLH / RTD
306* C  YLVLHR = VEHICLE PITCH ANGLE FROM LOCAL VERTICAL IN RADIAN$.
307* C  (ZYX EULER) YAW-PITCH SEQUENCE
308* C  ZLVLHR = ZLVLH / RTD
309* C  ZLVLHR = VEHICLE YAW ANGLE FROM LOCAL VERTICAL IN RADIAN$.
310* C  (ZYX EULER) YAW-PITCH SEQUENCE
311* C  XLVLHR = XLVLH / RTD
312* C  XLVLHR = VEHICLE ROLL ANGLE FROM LOCAL VERTICAL IN RADIAN$.
313* C  (ZYX EULER) YAW-PITCH SEQUENCE
314* C  PTSUNR = PTCOSUN / RTD
315* C  PTSUNR = LOOK ANGLE THETA FROM VEHICLE TO SUN (YAW-PITCH SEQUENCE)
316* C  YWSUNR = YAWSUN / RTD
317* C  YWSUNR = LOOK ANGLE PHI FROM VEHICLE TO SUN (YAW-PITCH SEQUENCE)
318* C  PTCOER = PTCOE / RTD
319* C  PTCOER = LOOK ANGLE THETA FROM VEHICLE TO CENTER OF EARTH.
320* C  YWCOER = YAWCOE / RTD
321* C  YWCOER = LOOK ANGLE PHI FROM VEHICLE TO CENTER OF EARTH.

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322*   C      SPARE - MATRIX FOR SPARE WORDS TO BE RESERVED FOR FUTURE USE
323*   C
324*   C THE ATTITUDE TAPE RECORDS ARE WRITTEN BELOW
325*   C
326*   WRITE (ITAPE) RECFLG,PRTFLG,YEAR,MONTH,GDAY,
327*   *GHR,GMIN,GSEC,GET,(RKMDP(I),I=1,3),(VKMDP(I),I=1,3),(DV(I),I=1,3),
328*   *YIMUDP,ZIMUDP,XIMUDP,VMAGKM,DECGCN,RASGCN,VMAGKM,GAMI,PSI,
329*   *VALTKM,VEHCDL,VEHLON,VELREL,GAMREL,PSIREL,GRAS,RFV,AKM,E,RADINC,
330*   *ASCNR,ARGPR,TRUEN,PERIOD,DYNPR,DRAGKG,LIFTKG,WGTKG,THRKG,I&PSEC,
331*   *EGRVAV,SHARON,(ATTECI(I,J),J=1,3),(ATTECI(I,J),J=1,3),
332*   *IATTECI(I,J),J=1,3,YLVLHR,ZLVLHR,XLVLHR,ALSR,BTSR,GMSR,PTSUNR,
333*   *YWSUNR,PTCOER,YWCOER,(RSUNKM(I),I=1,3),(SPARE(I),I=1,14)
334* 288 CONTINUE
335*   C
336*   IF (JPCARD.EQ.1) GO TO 290
337*   NUNCOM = 1
338* 290 CONTINUE
339*   IF (NUMLIN + NUNCOM) LE MAXLIN GO TO 300
340*   WRITE(6,950)
341* 950 FORMAT(1H1,43X,'SHUTTLE ATTITUDE AND POINTING TIME LINE')
342*   * 90X,'LOOK ANGLES, LOOK ANGLES, LOOK ANGLES'
343*   * 34X,'VEHICLE POSITION LVLH ATTITUDE',7X,'ECI ATTITUDE',6X,
344*   * 'TO SUN',6X,'TO EARTH',8X,'TO TARGET',5X,'GET',
345*   * 25X,'ALT',3X,'LAT',4X,'LON',4X,'YAW PITCH ROLL YAW PITCH',
346*   * 'ROLL YAW PITCH YAW PITCH YAW PITCH'
347*   * 1X,'HRS MN SECS',7X,'EVENT',9X,'N MI DEG',4X,'DFG DEG DEG',
348*   * 3X,'DEG DEG DEG DEG DEG DEG DEG DEG DEG DEG',
349*   * 3X,'DEG DEG')
350*   NUMLIN = 7
351* 300 CONTINUE
352*   VARR2 = VARR2
353*   C THE ATTITUDE AND POINTING TABLE IS WRITTEN BELOW.
354*   C
355*   IF (JPCARD.EQ.0) GO TO 305
356*   IF (IPTFLG.EQ.0) WRITE(6,960) IHT,IMT,ST,(NEVENT(J),J=1,3),
357*   * ALT,GDLAT,SCLON,ZLVLH,YLVLH,XLVLH,ZIMU,YIMU,XIMU,YAWSUN,
358*   * PTCSUN,YAWCOE,PTCCOE
359* 960 FORMAT(1X,I1,1X,I2,1X,F4.1,1X,3(A1),2(1X,F5.1),1X,F6.1,
360*   * 2(1X,3(1X,F5.1)),2(1X,2(1X,F5.1)))
361*   CALL TORSIF(ORR)
362*   IF (WRTFLG) 106,303,303
363* 303 CALL SWITCH
364*   IF (SBQANT(2) - SBQANT(1)) 301,306,301
365* 301 NCOUNT = NCOUNT + 1
366*   GOTO302
367* 304 CALL SWPRNT
368*   SBQANT(1) = SBQANT(2)
369*   GOTO306
370* 302 IF (IPTFLG.EQ.0) WRITE(6,970) IHT,IMT,ST,(NEVENT(J),J=1,3),
371*   * ALT,GDLAT,SCLON,ZLVLH,YLVLH,XLVLH,ZIMU,YIMU,XIMU,YAWSUN,
372*   * PTCSUN,YAWCOE,PTCCOE,NAMTAR,YAWTAR,PTCTAR,
373* 970 FORMAT(1X,I1,1X,I2,1X,F4.1,1X,3(A1),2(1X,F5.1),1X,F6.1,
374*   * 2(1X,3(1X,F5.1)),2(1X,2(1X,F5.1)),2X,A4,2(1X,F5.1))
375*   GOTO304
376* 305 CONTINUE
377*   IF (IPTFLG.EQ.0) WRITE(6,960) IHT,IMT,ST,NCARD,NCARD,NCARD,
378*   * ALT,GDLAT,SCLON,ZLVLH,YLVLH,XLVLH,ZIMU,YIMU,XIMU,YAWSUN,
379*   * PTCSUN,YAWCOE,PTCCOE

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380*      CALL TDRSIF(ORR)
381*      IF(WRTFLG .EQ. 307,307)
382*      307 CALL SWITCH
383*      IF(SBOANT(2) = SBOANT(1)) 308,306,308
384*      308 NCOUNT = NCRUNT + 1
385*      GOTO309
386*      311 CALL SWPRNT
387*      SBOANT(1) = SBOANT(2)
388*      GOTO306
389*      309 IF(PTFLG .NE. 0) WRITE(6,960) IHT,IMT,SY,NCARD,NCARD,NCARD,
390*      *ALT,GDLAT,SCLON,ZLVLH,YLVLH,XLVLH,ZIMU,YIMU,XIMU,VAWSUN,
391*      *PTCSUN,YAWCOE,PTCCOE,NAMTAR,YAWTAR,PTCTAR
392*      GOTO311
393*      306 CONTINUE
394*      IF(NUMCOM .EQ. 1) GO TO 350
395*      DO 320 J=2,NUMCOM
396*      IJ = (J-1)*3
397*      WRITE(6,980) NEVENT(IJ+1),NEVENT(IJ+2),NEVENT(IJ+3)
398*      980 FORMAT(13X,3(A2))
399*      320 CONTINUE
400*      C
401*      350 CONTINUE
402*      GETWRT = IHR + MN/60.0 + SECS/3600.0
403*      C THE DESCRIPTOR TAPE IS WRITTEN BELOW.
404*      C
405*      IF(IUWRT .GT. 0) WRITE(IUWRT) GETWRT,ATTSYS,IMUNUM,IVMS,
406*      * ANG1,ANG2,ANG3, IHOLD,XRATE,YRATE,ZRATE,PTFLG,ALIGN,INSTID,
407*      * IANG1,IANG2,IYPE1,IDEF1,VARI,VAR2,VAR3,ITARG1,VAR4,IDIN2,
408*      * IANG12,IANG22,IYPE2,IDEF2,VARI2,VAR22,VAR32,ITARG2,VAR42,
409*      * (NEVENT(J),J=1,12),NAMTAR
410*      C
411*      NUMLIN = NUMLIN + NUMCOM
412*      IF (GMTINT .GT. 1.E+4) GO TO 400
413*      IF (IBLANK .NE. 1) CALL CRDFIL(3)
414*      IBLANK = 0
415*      IF(IERR .EQ. 0) GO TO 100
416*      IF(IUWRT .GT. 0) ENDFILE IUWRT
417*      STOP
418*      400 CONTINUE
419*      GMT = GMT + GMTINT/60.000
420*      IF (JPCARD .EQ. 0) GO TO 450
421*      CALL CRDFIL(1)
422*      IF (ATTSYS .NE. NBLANK) CALL CRDFIL(3)
423*      C THE CALL TO GETDES OBTAINS THE NEXT EVENT SET
424*      C
425*      CALL GETDES,IUREAD,ISTOP)
426*      C
427*      IF (ITAPE .GT. 0 .AND. IERR .LT. 0) RECFLG = 1.0
428*      IF(RECFLG .LT. 0) WRITE(ITAPE) RECFLG,PTFLG,YEAR,GMONTH,GDAY,
429*      *GHR,GMIN,GSFC,DET,(RKMDP(I),I=1,3),(VKNDP(I),I=1,3),(DV(I),I=1,3),
430*      *YIMUDP,ZIMUDP,XIMUDP,RMAGKM,DECGCN,RASGCM,VMAGKM,GAMI,PSII,
431*      *VALTKM,VEHGD,VELON,VELREL,GAMREL,PSIREL,GRAS,RV,AKM,E,RADINC,
432*      *ASCNR,ARGPR,TRUEAN,PERIOD,DYNPR,DRAGK,LIFTKG,WTKG,THRKG,PCPSEC,
433*      *EGRAVC,SHADOW,(ATTECI(I,J),J=1,3),(ATTECI(7,J),J=1,3),
434*      * (ATTECI(3,J),J=1,3),YLVLHR,ZLVLHR,XLVLHR,ALSR,BTSR,GMSR,PTSUNR,
435*      *YWSUNR,PTCOER,YWCOER,(RSUNKM(I),I=1,3),(SPARE(I),I=1,14)
436*      IF (RECFLG .LT. 0) ENDFILE ITAPE
437*      IF (IERR .NE. 0) STOP

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```
438* CALL CRDFIL(I)
439* JPCARD = 0
440* TNEXT = IHR + MN/60 + SECS/3600
441* GMTNXT = TNEXT + BASTIM
442* 450 CONTINUE
443* IF ( (GMTNXT-GMT) .LE. 1.E-5 ) CALL CRDFIL(I)
444* IF ( (GMTNXT-GMT) .LE. 1.E-5 ) GO TO 120
445* ATTSYS = NBLANK
446* ALIGN = 0,
447* IF ( IERR .EQ. 0 ) GO TO 150
448* STOP
449* END
```

END OF COMPILATION: NO DIAGNOSTICS.
@ MAP,G SAPTP,SAPTP

SAPT PROGRAM (CONTINUED)

APPENDIX II

```
1*          SUBROUTINE SWITCH
2*          COMMON/ANGLES/,YAWTAR,PTCTAR,SBQANT(2),NCOUNT

3*          IF(PTCTAR .EQ. 0) PTCTAR=360.
4*          IF(PTCTAR .GT. 0 .AND. PTCTAR .LE. 90 ) GOTO10
5*          IF(PTCTAR .GT. 90 .AND. PTCTAR .LE. 180) GOTO20
6*          IF(PTCTAR .GT. 180 .AND. PTCTAR .LE. 270) GOTO30
7*          IF(PTCTAR .GT. 270 .AND. PTCTAR .LE. 360) GOTO40
8*          10 IF(180. - YAWTAR) 17,15,15
9*          15 SBQANT(2) = 1.0
10*         GOTO100
11*         17 SBQANT(2) = 2.0
12*         GOTO100
13*         20 IF(180. - YAWTAR) 15,17,17
14*         30 IF(180. - YAWTAR) 37,35,35
15*         35 SBQANT(2) = 4.0
16*         GOTO100
17*         37 SBQANT(2) = 3.0
18*         GOTO100
19*         40 IF(180. - YAWTAR) 35,37,37
20*         100 RETURN
21*         END
```

SUBROUTINE SWITCH

APPENDIX III


```
1*      SUBROUTINE TDRSIF(ORR)
2*      COMMON /ALONG/ SCLON,VARR2,WRTFLG

3*      IF(ORR .EQ. 1) GOTO35

4*      IF(ORR .EQ. 2) GOTO40

5*      IF(ORR .EQ. 0) GOTO48
6*      35 VARR2 = -41.0

7*      IF(VARR2 .EQ. -41.0) GOTO60
8*      48 VARR2 = -41.0
9*      ORR = ORR + 2
10*     GOTO50
11*     40 VARR2 = -171.0
12*     ORR = ORR - 2
13*     GOTO50
14*     60 ORR = ORR + 1
15*     50 IF(SCLON .GT. 106. .AND. SCLON .LE. 74.) GOTO 10
16*     GOTO20
17*     10 IF(VARR2 .LT. -170. .AND. VARR2 .GT. -172.) WRTFLG = -1.
18*     IF(VARR2 .LT. -40. .AND. VARR2 .GT. -42.) WRTFLG = 1.
19*     GOTO30
20*     20 IF(VARR2 .LT. -40. .AND. VARR2 .GT. -42.) WRTFLG = -1.
21*     IF(VARR2 .LT. -170. .AND. VARR2 .GT. -172.) WRTFLG = 1.
22*     30 RETURN
23*     END
```

SUBROUTINE TDRSIF

APPENDIX IV

```
1*      SUBROUTINE SWPRNT
2*      COMMON/ANGLES/ YAWTAR,PTCTAR,SBQANT(2),NCOUNT
3*      IANT=SBQANT(2)
4*      GO TO(10,20,30,40),IANT
5*      10 PRINT 11, NCOUNT
6*      11 FORMAT(14X,'SBQANT IS UPPER RT,'5X',15,' SWITCH POS CHANGES')
7*      GOTO50
8*      20 PRINT 21, NCOUNT
9*      21 FORMAT(14X,'SBQANT IS UPPER LT,'5X',15,' SWITCH POS CHANGES')
10*     GOTO50
11*     30 PRINT 31, NCOUNT
12*     31 FORMAT(14X,'SBQANT IS LOWER RT,'5X',15,' SWITCH POS CHANGES')
13*     GOTO50
14*     40 PRINT 41, NCOUNT
15*     41 FORMAT(14X,'SBQANT IS LOWER LT,'5X',15,' SWITCH POS CHANGES')
16*     50 RETURN
17*     END
```

SUBROUTINE SWPRNT

APPENDIX V

Z RUN MO0177,EE3,BEIA,MO45,C,10,5

ASG S=V06283

ASG F=V05655

ASG K=V04866

XQT CIR

TRN F,S,K

IN S

IN S

TRLS

XQT SAPT

1985 12 01 13 59 59.9

TORS -41

50 00 0.0 LVLH 1 321 0.0 0.0 0.0 .50 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0

TORS -171

50 0 0.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0

TORS -41

50 0 30.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0

TORS -171

50 0 30.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0

TORS -41

50 1 0.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0

TORS -171

50 1 0.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0

TORS -41

50 1 30.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0

TORS -171

50 1 30.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0

TORS -41

50 2 0.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0

TORS -171

50 2 0.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0

TORS -41

50 2 30.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0

TORS -171

50 2 30.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0

TORS -41

50 3 0.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0

TORS -171

50 3 0.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0

TORS -41

50 3 30.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0

TORS -171

50 3 30.0 1 321 1

2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0

RUN DECK FOR SAPT/SWITCH/SWPRNT/TDRSIF

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

```
-----  
TDRS -41  
50 4 .0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0  
TDRS -171  
50 4 30.0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0  
TDRS -41  
50 4 30.0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0  
TDRS -171  
50 4 30.0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0  
TDRS -41  
50 5 .0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0  
TDRS -171  
50 5 .0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0  
TDRS -41  
50 5 30.0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0  
TDRS -171  
50 5 30.0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0  
TDRS -41  
50 6 .0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0  
TDRS -171  
50 6 .0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0  
TDRS -41  
50 6 30.0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61074 -41.0 0.0  
TDRS -171  
50 6 30.0 1 321 1  
2 SEXTAN 0.0 147.5 G GRDTGT 5.61374 -171.0 0.0  
-----  
END CARD  
END
```

RUN DECK FOR SAPT/SWITCH/SWPRNT/TDRSIF
