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(NASA-TM-X-73479) HOT CORROSION STUDIES OF N77-10183 FOUR NICKEL-BASE SUPERALLOYS: B-1900, NASA-TRW VIA, 713C AND IN738 (NASA) 11 p HC A02/MF A01 CSCL 11F Unclas G3/26 08915

HOT CORROSION STUDIES OF FOUR NICKEL-BASE SUPERALLOYS: B-1900, NASA-TRW VIA, 713C and IN738

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ABSTRACT

The susceptibility to hot corrosion of four nickelbase superalloys has been studied at 900° and 1000°C in one atmosphere of slowly flowing oxygen. Hot corrosion was induced by coating the samples with known doses of Na₂SO₄ and oxidizing the coated samples isothermally on a sensitive microbalance. In general, the order of susceptibility found was: B-1900>713C>NASA-TRW VIA and > IN738. This order corresponds to the order of decreasing molybdenum content of the alloys. Chemical evidence for B-1900 indicates that hot corrosion is instigated by acid fluxing of the protective Al₂O₃ coating by MoO₂.

INTRODUCTION

The continually increasing performance demanded from gas turbine engines requires ever increasing operating temperatures, which in turn, requires alloys with increased high temperature strength and corrosion resistance. Higher strength has been achieved in the nickel-base alloys used in turbine engines by increasing the amount of **7**-forming elements, aluminum and titanium. This trend has provided alloys with good oxidation resistance but with decreased resistance to hot corrosion.

Most nickel-base superalloys are susceptible to hot corrosion attack to some extent. The hot corrosion phenomenon is known to be very complex, depending on many factors: temperature, composition of alloy, cyclic conditions, impurities in and pressure of air ingested into engine, and impurities in the fuel burned by the engine. Because of the complexity of the problem, research has been concerned mainly with phenomenalogical testing of alloys. Research directed toward an understanding of the reaction mechanisms has been less prevalent, and as a result, a complete fundamental characterization of hot-corrosion is still lacking. An exception to this trend has been the work of Goebel, Pettit, and Goward (Ref. 1) and of Bornstein, DeCrescente, and Roth (Ref. 2).

As part of a larger program aimed at understanding the chemical mechanisms involved in hot corrosion, we have conducted a laboratory-test study involving isothermal, thermogravimetric rate determinations of the hot corrosion of four superalloys: NASA-TRW VIA, B-1900, Alloy 713C, and IN-738. Hot corrosion was induced by coating the samples with measured doses of Na₂SO₄. Tests were conducted at 900° and 1000°C in one atmosphere of² slowly flowing oxygen. As a basis for comparison, we have also determined the corrosion rates in simple oxidation.

Hot corrosion of superalloys usually displays an induction period before enhanced or catastrophic oxidation begins. This is a period of little or no weight gain; in this paper, less than 0.3 mg cm^{-2} . The length of the induction period is a function of the temperature, the composition of the alloy, the concentration of Na₂SO₄, and the preoxidation treatment of the sample. We have been investigating the processes operative during and just after the induction period for samples of the four alloys that were preoxidized for 100 hours. The results obtained for B-1900 at 900°C with a coating of 3 mg cm⁻² Na₂SO₄ are discussed in terms of the "acidic fluxing" mechanism of hot corrosion (Ref. 1).

EXPERIMENTAL

The alloy samples were obtained from commercial sources and were given conventional heat treatments.

The samples were $0.2 \times 1.0 \times 2.5$ cm with a hangdown hole in one end. The surfaces were glass-bead blasted to give uniform matte surfaces, and were cleaned ultrasonically in detergent, acetone, and ethanol.

Hot corrosion was induced by coating the samples with Na₂SO₄ prior to oxidation as described by Bornstein et al. (Ref. 3). Two doses were used: 2/3 and 3 mg cm⁻².

Isothermal tests were performed at 900° and 1000°C in slowly flowing oxygen at one atmosphere pressure on both unoxidized

and preoxidized samples. The direction of flow of oxygen was downward and the speed was 20 cm min⁻¹. Continuous gravimetric measurements were made using a Cahn R-100 microbalance. Corrections in the weight change were made for the buoyancy effect. Exposure times extended in some cases to over 400 hours.

In the induction period studies, hot corrosion was induced with coatings of 3 mg cm⁻² at 900°C using samples that had been preoxidized for 100 hours at 900°C. Samples were run for set periods of time, and, after cooling, were water washed, usually for 15 minutes in warm water. The resulting solutions were analyzed for SO_h^- and metal cations and anions.

The morphology and composition of all oxide scales were examined by X-ray diffraction, light microscope, electron microprobe, SEM, and ESCA.

RESULTS AND DISCUSSION

<u>Straight Oxidation Studies</u>. The oxide scales formed on complex nickel-base superalloys are a complicated mixture of several individual oxides, the nature of which depends on many factors. However, for simplification, the alloys are often classified as "alumina formers" or "chromia formers." The alumina formers are characterized by a nonparabolic, nearly logarithmic, rate law with weight gains of the order of 0.2 to 0.5 mg cm⁻². The scale is composed of a thin, continuous layer of Al₂O₃ adjacent to the metal substrate. The chromia formers are characterized by a parabolic, or paralinear, rate law with weight gains of the order of 1 to 5 mg cm⁻² after 100 hours at 900° or 1000°C. The scale is characterized by a predominant Cr_2O_3 layer, usually with a region of internal oxidation of aluminum just beneath.

Our studies have shown that, in oxygen, B-1900 and VIA are primarily alumina formers, while 713C and 738 are chromia formers. This is unusual for 713C as it behaves as an alumina former when oxidized in air (Ref. 4).

<u>Comparative Hot Corrosion Studies</u>. Hot corrosion runs were conducted for all four alloys at 900 and 1000 °C with two doses of Na_2SO_1 : 2/3 and 3 mg cm². Both unoxidized and preoxidized

REPRODUCIBILITY OF T ORIGINAL PAGE IS POO (100 hrs. at temperature) samples were studied.

In Figures 1 and 2_{12} we present the results obtained at 900°C for the 2/3 and 3 mg cm² doses, respectively. The reproducibility of the data was roughly $\pm 15\%$ for duplicate samples. The induction period is not so obvious in Figures 1 and 2 because of the relatively insensitive time scales. Nevertheless, the onset of hot corrosion in all four alloys was preceded by an induction period. The induction period was generally followed by a period of rapidly accelerating weight gain that either consumed the entire sample: catastrophic oxidation; or transformed into a period of decelerating weight gain resulting in final cessation of oxidation: enhanced oxidation. Generally, we found enhanced oxidation with the 2/3 mg cm² dose and catastrophic oxidation with the 3 mg cm² dose.

The order of decreasing susceptibility to hot corrosion shown in Figures 1 and 2 is: B-1900 > 713C > VIA > 738, and this order was generally the same for all our results regardless of dose of Na₂SO₄ or temperature. It should be noted that 713C, a chromia former under the conditions of our experiments, is slightly more susceptible to hot corrosion than is VIA, an alumina former. In fact, the order of decreasing susceptibility is not the order of increasing chromium content, but is the order of decreasing molybdenum content: B-1900, 3.6; 713C, 2.4; VIA, 1.3; 738, 1.0 (in atomic percent).

It is noteworthy that 738 appears to be hot corrosion resistant for a considerable time, but after 200 hours undergoes very rapid catastrophic oxidation for the 3 mg cm⁻² dose (Fig. 2). In effect, it appears to have a very long induction period.

Preoxidation of the samples for 100 hours at temperature before hot corrosion had the effect of increasing the induction period markedly at 900°C, but only slightly at 1000°C. For example, the induction period at 900°C for B-1900 was extended from 1/2 to 6 hrs. for the 2/3 mg cm⁻² dose, and from $2\frac{1}{2}$ to $7\frac{1}{2}$ hrs. for the 3 mg cm⁻² dose; for 738 at 900°C and 3 mg cm⁻² dose, the induction period was greater than 450 hrs., the length of the test. At 1000°C, the induction period is a much less important feature of the hot corrosion curve being only 10 to 15 minutes long. It was found that though the induction period is increased by preoxidizing the samples, after the onset of hot corrosion, the rate of corrosion is the same as for nonpreoxidized samples; i.e., the hot corrosion curve is merely displaced in time and the extent of corrosion is the same for either type sample.

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The oxide scales formed during hot ocrrosion usually peeled off upon cooling to room temperature. They were composed of all the oxides of the metals in the alloy, NiO, of course, being the most prevalent. One combination oxide, NiMoO₁, was especially prominent in all four alloys, but especially in B-1900 and 713C. Interestingly, the oxide on VIA contained considerable Na(Ta, W)O₂.

<u>Induction Period Studies</u>. Samples of B-1900 preoxidized at 900°C for 100 hrs. were subjected to hot corrosion at 900°C by coating with 3 mg cm⁻² Na₂SO₁. Runs were continued for set periods of time as depicted by the arrows on the hot corrosion curve shown in Figure 3. The induction period in these experiments was roughly four hours. The actual value for individual samples varied, but all samples were normalized to a four hour induction period by matching the weight gain with the curve depicted in Figure 3.

The analytical results of the water soluble elements and SO_4 obtained from washing the samples are presented in Figure 4. We have plotted the amount of each element in milligrams against the length of time the sample underwent hot corrosion. The amount of SO_4 and Na originally applied to the samples is given at zero time. The total Na₂SO₄ was: 3 mg cm⁻² x 7 cm⁻² = 21 mg.

The sodium is completely recovered in all samples within the recovery ability of the simple water-washing process; some Na is imbedded in scale and is not dissolved. The $SO_4^{=}$ disappears rapidly, just at the end of the induction period. Soluble aluminum appears just after the induction period, indicating that the layer of Al₂O₃ on the alloy that protects it from oxidation has been dissolved. The soluble aluminum disappears for longer time periods indicating reformation of Al₂O₃, but not as an intact layer. Soluble chronium, which was not included in Figure 4 for sake of clarity, behaves in a fashion similar to aluminum, but reaches measurable values earlier in the induction period attaining a maximum of 2.2 mg. On the other hand, soluble nickel (also not included in Figure 4) was not observed until after the aluminum had maximized. It attained a value of 2.5 mg which

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOL tailed off gradually to a value of 1 mg.

The most striking feature of the results is the large amount of soluble molybdenum that appears just at the end of the induction period and beyond. At first thought, one might not expect soluble molybdenum to exceed that required to tie up the available Na as Na₂MoO₄. However, it must be remembered that MoO₂ is soluble in water to the extent of 2 g in 100 ml water at 70°C. This does not detract from the fact that considerable molybdenum has been oxidized in the early stages of the hot corrosion process.

Elemental X-ray micrographs of cross-sections of these hotcorroded samples were made. In Figure 5, we show the micrographs for several elements in the oxide scale of the $4\frac{1}{2}$ hr. sample; i.e., for a time just after the induction period, at the beginning of the period of rapidly accelerating oxidation. The protective Al₂O₂ layer has been dissolved and aluminum is dispersed throughout the oxide scale. The most unusual feature of these data is the position of the molybdenum concentrated along the oxide-alloy interface. This segregation is prominent in long time samples also and has been reported previously by Goebel, Pettit, and Goward (Ref. 1) and by Bourhis and St. John (Ref. 5).

<u>Mechanistic Implications</u>. The segregation of molybdenum along the alloy-oxide interface observed in the samples undergoing hot corrosion coupled with the large quantity of molybdenum detected in our analysis indicate that B-1900 hot corrodes by an "acidic fluxing" mechanism (Ref. 1) in contrast to the "basic fluxing" mechanism (Ref. 6). The results of our Induction Period Studies can be explained by the following sequence of reactions.

Initially, the $\operatorname{Cr}_2^0_3$ on our preoxidized samples is oxidized to soluble Cr_h^{-1} :

$$\operatorname{Cr}_{2}_{3} + 3/2 \circ_{2} + 2\operatorname{So}_{4}^{=} \rightarrow 2\operatorname{Cr}_{4}^{=} + 2\operatorname{So}_{3}$$
 (1)

This reaction continues throughout the induction period and consumes at most a few milligrams of SO_{1} .

Concomittantly, M_{OO_3} in the oxide is dissolving in the Na_oSO₁, consuming it by the reaction

$$M_{00}_{3} + S_{4} \rightarrow M_{00}_{4} + S_{3}$$
 (2)

In addition, the quantity of MoO₁ increases rapidly due to further oxidation of molybdenum carbides³lying in the surface oxide (Ref. 1).

By the end of the induction period, the MoO₂ has locally lowered the oxide ion activity in the melt to the point where dissolution of Al_2O_2 by "acidic fluxing" can take place:

$$A1_{2}O_{3} + 3M_{0}O_{3} \rightarrow 2A1^{+3} + 3M_{0}O_{1_{4}}^{=}$$
 (3)

At this point, the molten phase can creep beneath the Al₀ layer and spread over the sample as observed by Goebel, Pettit, and Goward (Ref. 1) and by Bourhis and St. John (Ref. 5).

The nature of the molten phase at this time is probably a mixture of MoO₃ (mp: 795°C) and Na₂MoO₄ (mp: 687°C). Our results indicate that little or no molten Na SO₄ is present in contrast to the hypothesis of Goebel, Pettit? and Goward (Ref. 1). It should be noted that the atomic weight of molybdenum is the same as the molecular weight of SO₄. Therefore, there is sufficient oxidized molybdenum to consume all the sulfate by reaction (2): Figure 4. Actually, we were not removing all the oxidized molybdenum in the micrographs), so the amount of molybdenum availaable for reaction (2) was probably considerably more than the amount reported in Figure 4.

The self-sustaining feature of the hot corrosion reaction arises from the presence of the molten phase, $MoO_3(1) + Na_2MoO_4(1)$, beneath the oxide scale adjacent to the alloy surface. The oxidation propagates by a process similar to that described by Brenner (Ref. 7) for Fe-Cr-Mo and Fe-Ni-Mo alloys.

The self-sustaining feature may be compromised by the reaction:

 $MoO_{3}(1) + NiO \rightarrow NiMoO_{4}$ (4)

This process replaces the liquid MoO₂ with a solid phase and accounts for the prominence of NiMoO₁ in all the hot corrosion scales. For the small dose of Na₂SO₁ ($\epsilon/3$ mg cm⁻²) used in this

study, oxidation ceased due to reaction (4). For the large dose (3 mg cm⁻²), oxidation continued until the sample was consumed, probably because $Na_2MoO_4(1)$ formed a larger fraction of the molten phase, and removal of $MoO_3(1)$ by reaction (4) was not as critical as with the smaller dose.

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REFERENCES

- J. A. Goebel, F. S. Pettit, and G. W. Goward, Met. Trans. <u>4</u>, 261 (1973).
- N. S. Bornstein, M. A. DeCrescente, and H. A. Roth, Met. Trans. <u>4</u>, 1799 (1973).
- 3. N. S. Bornstein and M. A. DeCrescente, Met. Trans. 2, 1971 (1971).
- 4. C. A. Barrett, J. Santoro, and C. E. Lowell, NASA TN D-7484, Nov. (1973).
- 5. Y. Bourhis and C. St. John, Oxid. Met. 9, 507 (1975).
- N. S. Bornstein, M. A. DeCrescente, and H. A. Roth, Proceedings of the 1972 Tri-Service Conference on Corrosion, MCIC 73-19.
- 7. S. S. Brenner, J. Electrochem. Soc. 102, 16 (1955).



200

Fig. 3. - Typical hot corrosion curve for B-1900 preoxidized samples.



400

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periods of hot corrosion - B-1900 preoxidized samples, 900° C, 3 mg cm⁻² Na₂SO₄.

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