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Computation of Unsteady Transonic Flows Through Rotating and Stationary Cascades

III - Acoustic Far-Field Analysis

Simon Slutsky, Dietrich Fischer,
and John I. Erdos

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Simon Slutsky, Dietrich Fischer,
and John I. Erdos

Advanced Technology Laboratories, Inc.
Westbury, New York

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LIST OF SYMBOLS

A_n, B_n, C_n, \dots	Fourier integral coefficients, defined by Equations (36) and (37).
A, B, C, D	Characteristic points, defined in Figure (2).
a	Speed of sound; m/sec.
b	Stream sheet thickness, m.
C_v	Specific heat at constant volume; N·m/kg·K.
H_m, J_m, K_m	Duct response functions.
H	Absolute total enthalpy per unit mass; N·m/kg.
i	$\sqrt{-1}$
i, j	Unit vectors in axial and circumferential directions, respectively.
M	Mach number.
m	Meridional distance; m.
N	Number of grid points along interface of numerical near-field and acoustic far-field solutions.
N_i	Number of blades in the i^{th} row.
Q, R	Boundary conditions at interface of numerical near-field and acoustic far-field solutions.
p	Static pressure, or pressure perturbation; N/m ² .
r	Radial distance from axis of rotation; m.
S	Entropy per unit mass; N·m/kg·K.
S_n	Function defined by Equation (49).
t	Time; sec.
T	Temperature, K.
T_n	Function defined by Equation (61).
U_o	Reference velocity; m/sec.
\vec{V}	Absolute velocity or velocity perturbation vector; m/sec.
V_m	Meridional (axial) component of velocity or velocity perturbation; m/sec.
V_θ	Absolute circumferential component of velocity or velocity perturbation; m/sec.
x	Axial distance; m.
y	Circumferential distance; m.
Y	Fundamental wavelength of stage in circumferential direction; m.
α_n, β_n	Acoustic propagation coefficients, defined by Equations (47) and (54); m ⁻¹ .
$\delta(\tau)$	Dirac delta function (0 for $\tau \neq 0$; 1 for $\tau = 0$).
Δ	Difference operator.
$\Delta(\tau)$	Heaviside step function (0 for $\tau < 0$; 1 for $\tau \geq 0$).

LIST OF SYMBOLS (Continued)

γ	Ratio of specific heats.
Ω	Non-dimensional frequency, defined by Equation (57).
ω	Frequency; sec^{-1} .
ρ	Static density or density perturbation, kg/m^3 .
θ	Circumferential angle; radians.
ζ	Radial component of vorticity; sec^{-1} .
∇	Vector operator.

Subscripts:

a, b, c, o	Evaluated at points A, B, C, O.
o	Reference state.
∞	Infinity condition.
i	Inlet station.
d	Discharge station.
m, n	Indices, defined where used.
1	First blade row.
1	Irrotational component of velocity vector.
2	Second blade row.
2	Solenoidal component of velocity vector.

Superscripts:

\rightarrow	Vector quantity.
—	Time to frequency transform.
o	Spatial location to spatial harmonic transform.
'	Perturbation variable.

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INTRODUCTION

Under the conditions typically prevalent in highly loaded transonic fan compressor stages, the linearized, small-perturbation approximations to the equations of motion cannot be expected to be descriptive of the flow in the vicinity of the blades. Thus, recourse is made to the numerical solution of the complete nonlinear system of equations, as discussed in Reference (1) in connection with the blade-to-blade program. However, sufficiently far from the blade rows, the amplitude of the flow disturbances will decay to acoustic levels and the linearized, small-perturbation approximations will be descriptive of the far-field. Therefore, an intermediate region in which both analyses are valid should exist at some distance from the blades. The inlet and discharge stations of the blade-to-blade computational domain can serve as the interfaces between the near-field (numerical) and far-field (acoustic) analyses. The present far-field analysis is formulated with respect to an infinite duct model, namely, all outgoing waves should propagate without reflection. It differs, however, from conventional inlet duct analyses in that the signal may begin with an arbitrary transient, associated with the deviation of the assumed initial data in the near-field from the periodic solution which is sought as the asymptotic limit in time. Therefore, the acoustic analysis must recognize that a transient signal will occur during startup and that a simple harmonic time dependence, which is the usual basis of inlet duct acoustics, cannot be assumed. The analysis should allow the transient to radiate outward without reflection, and should be capable of identifying the attainment of a periodic solution by the growth of discrete harmonic components in the solution.

INTERFACE WITH NEAR-FIELD SOLUTION

The procedures for defining inlet and discharge stations in terms of axial boundaries of the computational domains shown in Figure (1) have been described in References (1) and (2). The present analysis pertains to the flow upstream of the inlet station and downstream of the discharge station, where the passage is assumed to be an annular segment of an infinite cylindrical duct. With respect to the analysis of Reference (1), the streamsheet is assumed to have constant radius, r , and thickness, b . However, the circumferential boundaries of the passage are not necessarily restricted to a single blade-to-blade passage

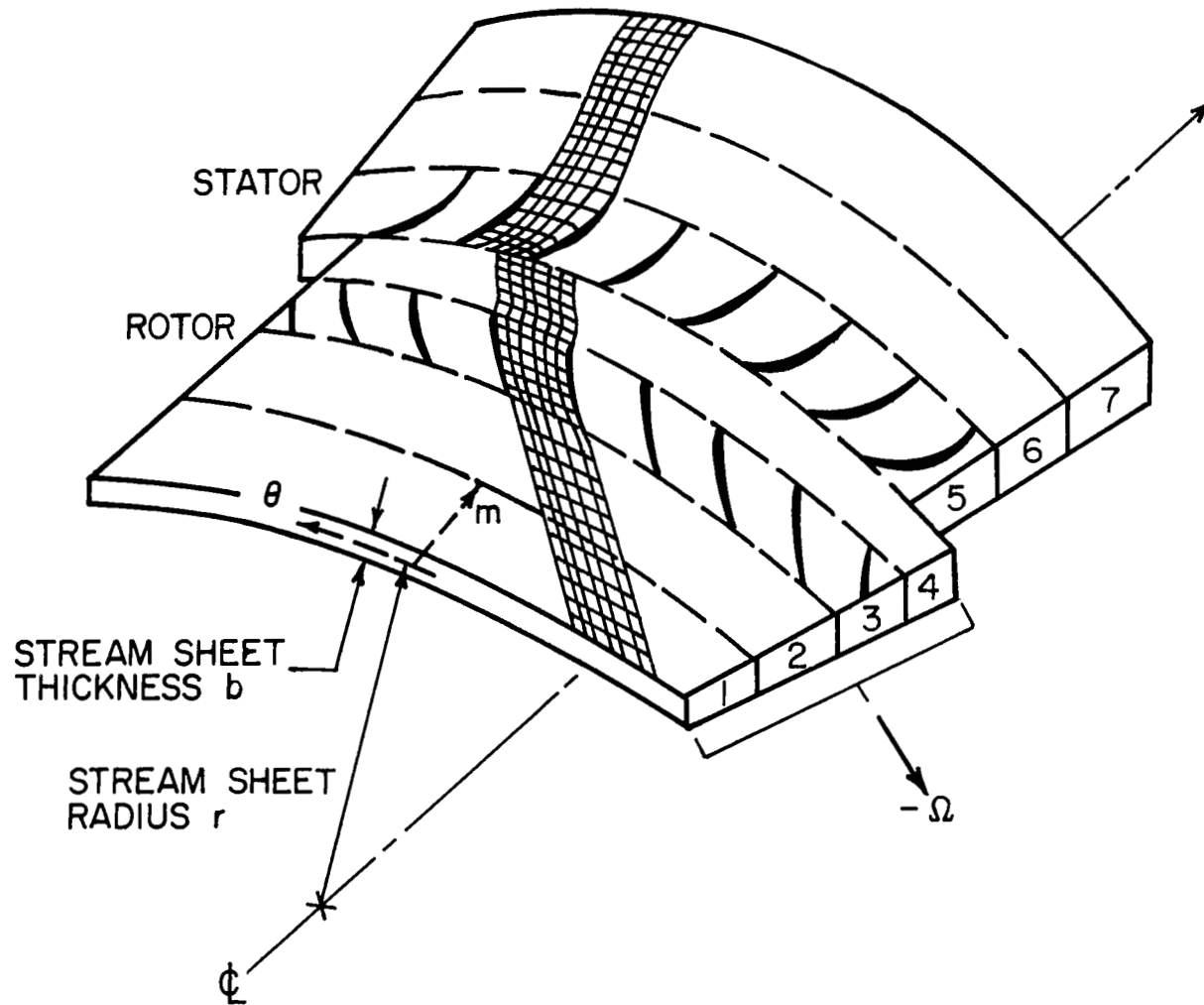


FIGURE 1. BLADE-TO-BLADE COORDINATE SYSTEM AND GRID NETWORK

(as in Reference 1), but should include the angular interval $2\pi/\Delta N$, where $\Delta N = |N_2 - N_1|$. (In the special case $\Delta N=0$, which is usually avoided in practice, use of a single blade-to-blade passage is again permissible.) Use of the present analysis with the computer code described in Reference (2) will, therefore, require additional storage and retrieval of data along the interfaces between the numerical near-field solution and the present acoustic far-field solution, in the same manner as is performed along the interface between domains 4 and 5 of the near-field solution (see Reference 2).

Derivation of the boundary conditions on the near-field (finite difference) solution, at arbitrary stations referred to as the inlet and discharge boundaries of the near field, was presented in Volume 1 (Reference 1). The acoustic analysis is formulated in the absolute frame of reference, and accordingly the boundary point analysis from Reference (1) will be restated here in absolute coordinates.

As discussed in Reference (1), the boundary point solution is obtained from a reference plane method-of-characteristics procedure.* A pair of (approximately) two-dimensional wave motion characteristics and a stream surface characteristic are identified at each grid point on the inlet and discharge boundaries, as shown in Figure (2). The compatibility relations,

$$\frac{dp}{dt} \pm \rho a \frac{dV_m}{dt} = -\gamma p \left(\frac{\partial V_\theta}{r \partial \theta} + \frac{V_m}{rb} \frac{drb}{dm} \right) \pm \rho a V_\theta^2 \left(\frac{1}{r} \frac{dr}{dm} \right) \quad (1)$$

apply on the wave motion characteristics,

$$\frac{dm}{V_m \pm a} = \frac{rd\theta}{V_\theta} = dt \quad (2)$$

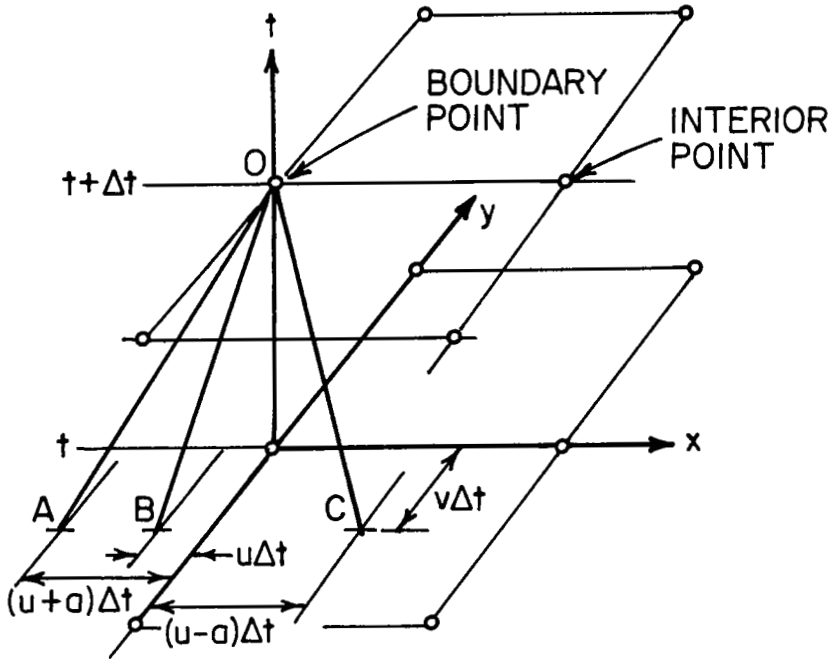
which are depicted by the lines AO and CO in Figure (2). The equation,

$$S = \text{constant} \quad (3)$$

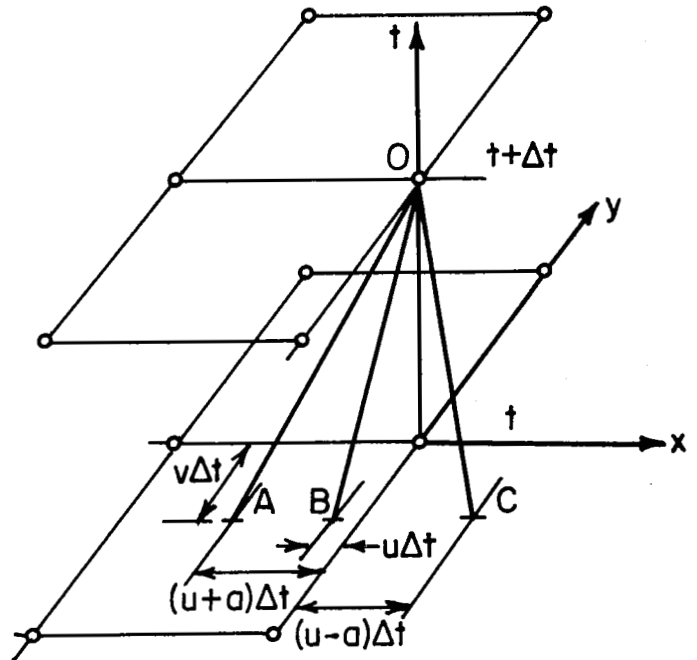
applies on the stream path characteristic,

$$\frac{dm}{V_m} = \frac{rd\theta}{V_\theta} = dt \quad (4)$$

*As pointed out in Reference (1), translation of the reference plane effectively transforms a rotating frame of reference back to an absolute frame. Thus the numerical results are independent of the coordinate system in which the equations are stated.



INLET STATION



DISCHARGE STATION

FIGURE 2. CHARACTERISTIC LINES AND GRID POINTS AT INLET AND DISCHARGE STATIONS

which is depicted by the line B0 in Figure (2). In addition, if the flow is isentropic

$$\zeta/\rho = \text{constant}$$

also applies on the stream path characteristic, B0. Finally, the circumferential momentum equation is written as:

$$\frac{\partial V_\theta}{\partial t} = -\frac{\partial H}{r\partial\theta} + T\frac{\partial S}{r\partial\theta} - V_m \zeta \quad (5)$$

The inlet flow is assumed to be isentropic and irrotational; thus:

$$S_i = 0, \quad \text{or} \quad p_i/\rho_i^\gamma = p_{-\infty}/\rho_{-\infty}^\gamma \quad (6)$$

and

$$\zeta_i = 0 \quad (7)$$

giving:

$$\left(\frac{\partial V_\theta}{\partial t}\right)_i = -\left(\frac{\partial H}{r\partial\theta}\right)_i \quad (8)$$

Introducing the small disturbance approximations,

$$p' = p - p_{-\infty} \ll p_{-\infty} \quad (9)$$

$$V_m' = V_m - V_{m_{-\infty}} \ll V_{m_{-\infty}} \quad (10)$$

$$V_\theta' = V_\theta \ll V_{m_{-\infty}} \quad (11)$$

etc.

the compatibility equation on the outgoing wave, C0, at the inlet becomes:

$$\begin{aligned} p_i' - \rho_{-\infty} a_{-\infty} V_{m_i}' &= p_c' - \rho_{-\infty} a_{-\infty} V_{m_c}' \\ &- \frac{1}{2} \left(\gamma p_{-\infty} \left(\frac{\partial V_\theta}{r\partial\theta}\right)_i + \gamma p_c \left(\frac{\partial V_\theta}{r\partial\theta} + \frac{V_m}{rb} \frac{drb}{dm}\right)_c \right) \Delta t \\ &- \left(\rho_{-\infty} a_{-\infty} V_{\theta_c}^2 \left(\frac{1}{r} \frac{dr}{dm}\right)_c \right) \Delta t \end{aligned} \quad (12)$$

where the subscript i denotes the inlet value at a new time, $t+\Delta t$, i.e., at point 0 in Figure (2), and subscript c denotes the value at point C. Recall that r and b are assumed constant for $m \leq m_i$. Since V_θ can be evaluated at all points 0 along the inlet, from integration of Equation (8), and all variables at point C can be obtained by interpolation of known (current) data in the near-field, the right-hand side can be regarded as a known function of (m_i, θ, t) , i.e.:

$$p'_i - \rho_{-\infty} a_{-\infty}^2 V'_{m_i} = \rho_{-\infty} a_{-\infty}^2 Q_i(m_i, \theta, t) \quad (13)$$

$Q_i(m_i, \theta, t)$ is taken as the boundary condition to be placed on the upstream far-field solution. As will be shown in the following section, this condition is sufficient to determine p'_i , and thus p_i . V'_{m_i} can then be obtained from Equation (13), and ρ_i from Equation (6). As indicated above, V_{θ_i} is obtained from Equation (8), completing the inlet solution at the new time $t+\Delta t$.

Similar procedures at the discharge boundary lead to the relations:

$$p_d/\rho_d^Y = p_b/\rho_b^Y \quad (14)$$

$$\left(\frac{\partial V_\theta}{\partial t}\right)_d = -\left(\frac{\partial H}{r \partial \theta} - \frac{T \partial S}{r \partial \theta} + V_m \zeta\right)_d \quad (15)$$

$$p'_d + \rho_\infty a_\infty^2 V'_{m_d} = \rho_\infty a_\infty^2 Q_d(m_d, \theta, t) \quad (16)$$

Equation (16) represents one boundary condition on the downstream far-field. Integration of Equation (15) provides a second boundary condition on the downstream far-field, which is required since it cannot be assumed to be irrotational.

$$V_{\theta_d} = a_\infty R_d(m_d, \theta, t) \quad (17)$$

Due to the finite work input by the rotor, the downstream reference conditions (ρ_∞, a_∞) are also not necessarily the same as the upstream reference conditions $(\rho_{-\infty}, a_{-\infty})$. In this case, the far-field analysis provides the value of p'_d and Equations (14), (16) and (17) complete the discharge station solution at the new time $t+\Delta t$.

With these preliminaries in hand, attention is now focussed on the acoustic far-field analysis. In the following discussion the primes on the perturbation variables will be dropped as the acoustic solution is understood to pertain only to perturbation quantities. Furthermore, the subscript $()_0$ will be used to denote the reference conditions for either boundary, that is, $x \rightarrow \pm \infty$, and all variables without subscript will refer to perturbations with respect to the reference state, for example, $p = p - p_0$ and $\vec{V} = \vec{V} - \vec{V}_0$.

FAR-FIELD ANALYSIS

The problem of interest is the acoustic radiation from the considered blade rows under steady-state operating conditions. However, development of the numerical solution for the near-field includes generation of a transient flow field associated with the deviation of the assumed initial conditions from a periodic solution, which is sought as an asymptotic limit in time. Therefore, analysis of the far-field must necessarily recognize that a transient signal will occur during startup and that a simple harmonic time dependence, which is the usual basis of engine acoustic analyses, cannot be assumed to be descriptive until the asymptotic limit is approached. The analysis should allow the transient to radiate outward, without reflection, and should be capable of identifying the attainment of a periodic solution by the growth of discrete harmonic components.

Use will be made of a harmonic type analysis to develop an acoustic formulation of the pressure field in which frequency effects are superimposed and the time history obtained by integration. This procedure leads to a convolution integral (i.e., a type of Duhamel integral) in the time domain, which is more directly compatible with the numerical data in the near-field than a result in the frequency domain. (The near-field data consists of values of the pressure, flow velocity, etc., at a discrete series of grid points and at fixed time intervals.)

Since the present effort is addressed toward a cascade formulation, the governing equations are written in a two-dimensional, Cartesian coordinate system. It is noted that the two-dimensional problem could be described by a solution of the wave equation alone, were it not for the fact that the time dependent force distribution on the blades is capable of producing a convected

vorticity field (as well as a corresponding entropy field, which, however, is not relevant to the present problem). It will be seen that the convected vorticity field does not contribute to the acoustic pressure field, per se, as distinguished from the irrotational component of the velocity field which is directly coupled with the acoustic pressure. Therefore, the velocity perturbation field downstream of the discharge boundary and upstream of the inlet boundary (see Figure 1) is characterized by the sum of an irrotational ($\nabla \times \vec{V} = 0$) velocity vector \vec{V}_1 and a solenoidal ($\nabla \cdot \vec{V} = 0$) velocity vector \vec{V}_2 , which satisfy the linearized conservation equations. Note that the inlet flow has been assumed to be irrotational, therefore, $\vec{V}_2 = 0$ in the upstream far-field; however, the analysis will be developed with respect to the more general case pertaining to the downstream far-field.

The linearized conservation equations for the perturbation variables can be stated as*

$$\frac{dp}{dt} + \rho_0 a_0^2 \nabla \cdot \vec{V} = 0 \quad (18)$$

$$\rho_0 \frac{d\vec{V}}{dt} + \nabla p = 0 \quad (19)$$

$$\frac{dp}{dt} - a_0^2 \frac{d\rho}{dt} = \frac{p_0}{C_v} \frac{dS}{dt} \quad (20)$$

where:

$$\vec{V} = \vec{V}_1 + \vec{V}_2 \quad (21)$$

$$\vec{V}_1 = \vec{i} v_{m_1} + \vec{j} v_{\theta_1} \quad (22)$$

$$\vec{V}_2 = \vec{i} v_{m_2} + \vec{j} v_{\theta_2} \quad (23)$$

$$\nabla \times \vec{V}_1 = 0 \quad (24)$$

$$\nabla \times \vec{V}_2 = \zeta \neq 0 \quad (25)$$

*These represent linearized versions of Equations (13), (14), (15) and (17) of Reference (1), with $r = \text{constant}$ and $b = \text{constant}$.

$$\nabla \cdot \vec{V}_2 = 0 \quad (26)$$

$$U_o = M_o a_o \quad (27)$$

and

$$\frac{d}{dt} = \frac{\partial}{\partial t} + U_o \frac{\partial}{\partial m} \quad (28)$$

Since the streamsheet has been assumed to be cylindrical in the far-field, with $b \ll r$, the conventional Cartesian coordinates (x,y) will be used in place of the meridional coordinates (m,θ) :

$$x = m \quad (29a)$$

$$y = r\theta \quad (29b)$$

Substituting Equation (26) into (18) gives:

$$\frac{dp}{dt} + \rho_o a_o^2 \nabla \cdot \vec{V}_1 = 0 \quad (30)$$

Substituting Equations (24) and (26) into the curl and divergence of Equation (19), respectively, gives:

$$\frac{d}{dt} (\nabla \cdot \vec{V}_1) + \frac{1}{\rho_o} \nabla^2 p = 0 \quad (31)$$

and:

$$\frac{d}{dt} (\nabla \times \vec{V}_2) = 0 \quad (32)$$

Equations (18) and (19) or (30) and (31) yield the wave equation for the pressure:

$$\frac{1}{a_o^2} \frac{d^2 p}{dt^2} - \nabla^2 p = 0 \quad (33)$$

Therefore, solutions for p are purely radiative. Since Equations (30) and (31) can be also combined to yield the wave equation, solutions for \vec{V}_1 are also purely radiative. However, Equation (32) infers that solutions for \vec{V}_2 are purely convective. Therefore:

$$\frac{d\vec{V}_2}{dt} = 0 \quad (34)$$

and:

$$\frac{d\vec{V}_1}{dt} + \frac{1}{\rho_0} \nabla p = 0 \quad (35)$$

Solutions to Equations (33), (34) and (35) can be expressed in terms of a Fourier integral representation:

$$\begin{Bmatrix} p(x,y,t) \\ v_{m_1}(x,y,t) \\ v_{\theta_1}(x,y,t) \end{Bmatrix} = \sum_n e^{-i\alpha_n y} \int_{-\infty}^{\infty} \begin{Bmatrix} \rho_0 a_0 A_n(\omega) \\ B_n(\omega) \\ C_n(\omega) \end{Bmatrix} a_0 e^{i(\omega t - \beta_n x)} \frac{d\omega}{2\pi} \quad (36)$$

$$\begin{Bmatrix} v_{m_2}(x,y,t) \\ v_{\theta_2}(x,y,t) \end{Bmatrix} = \sum_n e^{-i\alpha_n y} \int_{-\infty}^{\infty} \begin{Bmatrix} D_n(\omega) \\ E_n(\omega) \end{Bmatrix} a_0 e^{i\omega(t-x/U_0)} \frac{d\omega}{2\pi} \quad (37)$$

Consider first the downstream boundary on which (see Equations 16 and 17):

$$Q_d(y,t) = (p_d + \rho_0 a_0 v_{m_d}) / (\rho_0 a_0^2) \quad (38)$$

$$R_d(y,t) = v_{\theta_d} / a_0 \quad (39)$$

Thus, the boundary data can be expressed as:

$$\begin{Bmatrix} Q(y,t) \\ R(y,t) \end{Bmatrix} = \sum_n e^{-i\alpha_n y} \int_{-\infty}^{\infty} \begin{Bmatrix} \bar{Q}_n^0(\omega) \\ \bar{R}_n^0(\omega) \end{Bmatrix} e^{i\omega t} \frac{d\omega}{2\pi} \quad (40)$$

where $(\bar{\quad})$ denotes the time to frequency transform, as above, while $(\quad)^0$ indicates transformation from spatial location, y , to spatial harmonic, n , which will be discussed later (cf. Equation 55). (The subscript d has been dropped since the ensuing development can apply to either the inlet or discharge boundary.)

Substitution of the above integral forms into Equation (24), (26), (32), (34), (38) and (39) gives:

$$C_n = \frac{\alpha_n}{\beta_n} B_n \quad (41)$$

$$E_n = -\frac{\omega}{U_o \alpha_n} D_n \quad (42)$$

$$B_n = \frac{a_o \beta_n}{\omega - U_o \beta_n} A_n \quad (43)$$

$$A_n + B_n + D_n = \bar{Q}_n^o \quad (44)$$

$$C_n + E_n = \bar{R}_n^o \quad (45)$$

The value of A_n , and thus B_n , C_n , D_n and E_n , can thereby be expressed in terms of the transform of the boundary data, viz.:

$$A_n = \frac{\omega - U_o \beta_n}{\omega^2 + (a_o - U_o) \beta_n \omega + U_o a_o \alpha_o^2} [\omega \bar{Q}_n^o + U_o \alpha_n \bar{R}_n^o] \quad (46)$$

The propagation coefficient β_n is determined by substituting the integral relation for pressure (Equation 36) into the wave equation (Equation 33):

$$\beta_n = \frac{-\omega M_o \pm (\omega^2 - \alpha_n^2 a_o^2 (1 - M_o^2))^{\frac{1}{2}}}{a_o (1 - M_o^2)} \quad (47)$$

where the + sign refers to downstream propagation (from the discharge boundary) and the - sign to upstream propagation (from the inlet boundary).

Equations (46) can now be substituted into the corresponding integral relation (Equation 36) to obtain, after some manipulation, the pressure perturbation on the discharge boundary in terms of the specified boundary data.

$$p_d(y, t) = \frac{\rho_o a_o}{(1 - M_o^2)} \sum_n \frac{e^{-i \alpha_n y}}{\alpha_n a_o} \int_{-\infty}^{\infty} (\omega - S_n) \left(\frac{\omega}{\alpha_n a_o} \bar{Q}_n^o + M_o \bar{R}_n^o \right) e^{i \omega t} \frac{d\omega}{2\pi} \quad (48)$$

where

$$s_n = (\omega^2 - \alpha_n^2 a_o^2 (1-M_o^2))^{\frac{1}{2}} \quad (4)$$

Consider now the inlet boundary where $\vec{V}_2 = 0$. The boundary condition here is stated as (see Equation 13):

$$Q_i(y,t) = (p_i - \rho_o a_o v_{m_i}) / (\rho_o a_o^2) \quad (5)$$

Since $\vec{V}_2 = 0$, $D_n = E_n = 0$, and \bar{R}_n^o becomes part of the solution, i.e., it cannot be specified as a boundary condition, consistent with the near field analysis described in the previous section. Thus, A_n is only a function of Q_i , viz:

$$A_n = \frac{\omega - U_o \beta_n}{\omega - (a_o + U_o) \beta_n} \bar{Q}_n^o \quad (5)$$

where

$$Q_i(y,t) = \sum_n e^{-i\alpha_n y} \int_{-\infty}^{\infty} \bar{Q}_n^o(\omega) e^{i\omega t} \frac{d\omega}{2\pi} \quad (5)$$

Again, with some manipulation, an equation for the pressure perturbation on the inlet boundary is obtained in terms of the specified data:

$$p_i(y,t) = \frac{\rho_o a_o^2}{(1-M_o^2)(1+M_o)} \sum_n \frac{e^{-i\alpha_n y}}{\alpha_n^2 a_o^2} \int_{-\infty}^{\infty} (\omega - s_n) (\omega + M_o s_n) \bar{Q}_n^o e^{i\omega t} \frac{d\omega}{2\pi} \quad (5)$$

The selected representation of the solutions, Equations (36) and (37), and boundary conditions, Equations (40) and (52), as Fourier series in the y direction is appropriate for enforcement of the periodicity boundary condition per-

taining to spatial variations in this direction. The coefficient α_n is defined accordingly:

$$\alpha_n = \frac{2\pi n}{Y} = \frac{n\Delta N}{r} \quad (54)$$

where Y is the fundamental wave length of the stage (double cascade) configuration (i.e., $Y = 2\pi r/\Delta N$). In addition, the fact that the boundary data is specified at a discrete number of grid points, say N , on the boundaries implies that the Fourier series can only include N terms; $n=0, 1, 2, \dots, N-1$. Since the distance y to each point can be written as mY/N , where $m=0, 1, 2, \dots, N-1$ also, the Fourier series can be expressed in the standard Discrete Fourier Transform (DFT) notation:

$$\begin{pmatrix} Q(y,t) \\ R(y,t) \end{pmatrix} = \begin{pmatrix} Q_m(t) \\ R_m(t) \end{pmatrix} = \sum_{n=0}^{N-1} \begin{pmatrix} Q_n^o(t) \\ R_n^o(t) \end{pmatrix} e^{-\frac{2\pi i n m}{N}} \quad (55)$$

The inverse DFT is then:

$$\begin{pmatrix} Q_n^o(t) \\ R_n^o(t) \end{pmatrix} = \frac{1}{N} \sum_{m=0}^{N-1} \begin{pmatrix} Q_m(t) \\ R_m(t) \end{pmatrix} e^{2\pi i n m/N} \quad (56)$$

For computational purposes, the following non-dimensional parameters are evident in Equations (48) and (53):

$$\Omega = \omega/\alpha a_o \quad (57)$$

$$\tau = \alpha a_o t \quad (58)$$

where

$$\alpha = 2\pi/Y = \alpha_n/n \quad (59)$$

Then, on the discharge boundary:

$$p_n^o(\tau) = \frac{\rho_o a_o^2}{(1-M_o^2)n^2} \int_{-\infty}^{\infty} (\Omega - T_n) (\Omega Q_n^o(\Omega) + n M_o R_n^o(\Omega) e^{i\Omega\tau}) \frac{d\Omega}{2\pi} \quad (60)$$

where:

$$T_n = (\Omega^2 - (1-M_o^2)n^2)^{\frac{1}{2}} \quad (61)$$

$$Q_n^o(\Omega) = \alpha a_o Q_n^o(\omega) \quad (62)$$

$$R_n^o(\Omega) = \alpha a_o R_n^o(\omega) \quad (63)$$

and on the inlet boundary:

$$p_n^o(\tau) = \frac{\rho_o a_o^2}{(1-M_o^2)(1+M_o)n^2} \int_{-\infty}^{\infty} (\Omega - T_n) (\Omega + M_o T_n) Q_n^o(\Omega) e^{i\Omega\tau} \frac{d\Omega}{2\pi} \quad (64)$$

Then, on either boundary:

$$p_m(\tau) = \sum_{n=0}^{N-1} e^{-2\pi i n m / N} p_n^o(\tau) \quad (65)$$

The integrals of Equations (60) and (64) can be transformed back to the time domain by the following convolution. On the discharge boundary:

$$p_n^o(\tau) = \rho_o a_o^2 (H_n^o(\tau) * Q_n^o(\tau) + J_n^o(\tau) * R_n^o(\tau)) \quad (66)$$

where * denotes convolution in time. For $n \geq 1$:

$$H_n^o(\tau) = \frac{1}{n^2(1-M_o^2)} \int_{-\infty}^{\infty} \Omega (\Omega - T_n) e^{i\Omega\tau} \frac{d\Omega}{2\pi} \quad (67)$$

$$J_n^0(\tau) = \frac{1}{n(1-M_0^2)} \int_{-\infty}^{\infty} (\Omega - T_n) e^{i\Omega\tau} \frac{d\Omega}{2\pi} \quad (68)$$

and, for $n=0$:

$$H_0^0(\tau) = \frac{1}{2} \delta(\tau) \quad (69)$$

$$J_0^0(\tau) = 0 \quad (70)$$

$$p_0^0(\tau) = \frac{1}{2} \rho_0 a_0^2 Q_0^0(\tau) \quad (71)$$

where δ is the Dirac delta function.

Note that for $t \rightarrow 0$ with $n \neq 0$:

$$H_n^0(\tau) \rightarrow \frac{1}{2} \delta(\tau) \quad (72)$$

$$p_n^0(\tau) \rightarrow \frac{1}{2} \rho_0 a_0^2 Q_n^0(\tau) \quad (73)$$

On the inlet boundary:

$$p_n^0(\tau) = \rho_0 a_0^2 K_n^0(\tau) * p_n^0(\tau) \quad (74)$$

where, for $n \geq 1$

$$K_n^0(\tau) = \frac{1}{n^2 (1-M_0^2) (1+M_0^2)} \int_{-\infty}^{\infty} (\Omega - T_n) (\Omega + M_0 T_n) e^{i\Omega\tau} \frac{d\Omega}{2\pi} \quad (75)$$

and, for $n=0$:

$$k_0^o(\tau) = \frac{1}{2} \delta(\tau) \quad (76)$$

$$p_0^o(\tau) = \frac{1}{2} \rho_0 a_0^2 p_0^o(\tau) \quad (77)$$

while for $t \rightarrow 0$ with $n \neq 0$:

$$k_n^o(\tau) = \frac{1}{2} \delta(\tau) \quad (78)$$

$$p_n^o(\tau) = \frac{1}{2} \rho_0 a_0^2 p_n^o(\tau) \quad (79)$$

The integral in Equation (67) is evaluated by dividing the non-dimensional frequency range into three regions, viz: $-\infty$ to $-\Omega_0$, $-\Omega_0$ to Ω_0 , and Ω_0 to ∞ . In the middle region standard DFT techniques are applied, while in the outer regions, $\Omega \rightarrow \pm \infty$, the integral is accurately approximated in terms of an exponential integral of the form

$$\int_{-\Omega_0}^{+\infty} \frac{e^{i\Omega\tau}}{\Omega^\alpha} d\Omega \quad (80)$$

for which closed form expressions are available. The integral in Equation (68) can be expressed in terms of a Bessel function:

$$\int_{-\infty}^{\infty} e^{i\Omega\tau} (\Omega - (\Omega^2 - c^2)^{\frac{1}{2}}) \frac{d\Omega}{2\pi} = ic^2 \frac{J_1(c\tau)}{c\tau} \Delta(\tau) \quad (81)$$

where c denotes $n(1-M_0^2)^{\frac{1}{2}}$ and $\Delta(\tau)$ is the Heaviside step function. The evaluation of the integral in Equation (75) is carried out in the same fashion as that in Equation (67).

The desired solution for the pressure perturbation on the interface with near-field is finally accomplished by observing that Equation (65) can now be expressed in the form:

$$p_m(t) = \rho_o a_o^2 \sum_{n=0}^{N-1} e^{-2\pi i n m / N} \left\{ \begin{array}{l} H_n^o(t) * Q_n^o(t) + J_n^o(t) * R_n^o(t) \\ K_n^o(t) * Q_n^o(t) \end{array} \right\} \quad (82)$$

on the discharge and inlet boundaries respectively. Use of Equation (55) for $P_m(t)$, $Q_m(t)$ and $R_m(t)$ and similar DFT expansions for H_m , J_m and K_m then leads to the convolution:

$$p_m(t) = \frac{\rho_o a_o^2}{N} \sum_{n=0}^{N-1} \left\{ \begin{array}{l} H_n(t) * Q_{m-n}(t) + J_n(t) * R_{m-n}(t) \\ K_n(t) * Q_{m-n}(t) \end{array} \right\} \quad (83)$$

on the discharge and inlet boundaries respectively. Thus, a double convolution over both time and distance (in the y direction) is required. The functions* $Q_m(t)$ and $R_m(t)$, therefore, represent the point sources of time-varying strength which are aligned along the considered boundaries, with spatial resolutions consistent with the number of grid points specified.

The functions $H_m(t)$, $J_m(t)$ and $K_m(t)$ are the duct response functions, defined by:

*Bear in mind that $Q_m(t)$ denotes either $Q_i(y_m, t)$ at the inlet station or $Q_d(y_m, t)$ at the discharge station, while $R_m(t)$ denotes $R_d(y_m, t)$.

$$\begin{aligned}
\begin{pmatrix} H_m(t) \\ J_m(t) \\ K_m(t) \end{pmatrix} &= \sum_{n=0}^{N-1} \begin{pmatrix} H_n^O(t) \\ J_n^O(t) \\ K_n^O(t) \end{pmatrix} e^{-2\pi i n m / N} \\
&= \begin{pmatrix} \frac{1}{2} \delta(t) \\ 0 \\ \frac{1}{2} \delta(t) \end{pmatrix} + \sum_{n=1}^{N-1} \begin{pmatrix} H_n^O(t) \\ J_n^O(t) \\ K_n^O(t) \end{pmatrix} e^{-2\pi i n m / N}
\end{aligned} \tag{84}$$

Finally, it should be noted that only the real part of $p_m(t)$ is of interest. Equations (67), (68) and (75) infer that $H_n^O(\tau)$ and $K_n^O(\tau)$ are purely real functions and $J_n^O(\tau)$ is a pure imaginary function. The boundary data $Q_m(t)$ and $R_m(t)$ are obviously all real numbers. Therefore, it is concluded that

$$J_m(t) = \sum_n -i J_n^O(t) \sin \frac{2\pi n m}{N} = \sum_n L_n^O(t) \sin \frac{2\pi n m}{N} \tag{85}$$

where

$$L_n^O(\tau) = \frac{1}{n(1-M_0^2)} \int_{-\infty}^{\infty} (\Omega - T_n) e^{i\Omega\tau} \frac{d\Omega}{2\pi} \tag{86}$$

i.e., the solution can be carried out entirely in terms of real functions.

NUMERICAL EXAMPLE

A group of subroutines including an efficient Fast Fourier Transform routine has been developed, for use in conjunction with the blade-to-blade computer program B2DATL, described in Reference (2), to carry out the indicated convolutions required to solve Equation (83) numerically. Results have been thus far limited to test cases with a simple harmonic input signal. For

example, in one of the calculations the discharge station was assumed to be divided into 8 intervals covering a total circumferential distance of 0.1 m. The selected reference (average) Mach number of the discharge flow was 0.8. The flow was assumed to be irrotational so that only one input function $Q(y,t)$ was required, and the second function $R(y,t)$ could be considered as a response function (that is, it was calculated from $Q(y,t)$). The input function $Q(y,t) = \cos(\Omega\tau - 2\pi n/N)$, with $Q(y,t) = 0$ for $\tau < 0$, was selected for this case, where $\tau = 2\pi Na_0 t/Y$, ($n = 0, 1, 2, \dots, N-1$) $N = 8$, $Y = 0.1$ m., $a_0 = 10^3$ m/sec, and $\Omega = 1$. The input function Q and response function R at $n = 0$ are plotted in Figure (3). It should be noted that the response function is initially out of phase with the input function because of the assumption that $Q = 0$ for $\tau < 0$. However, the effect of the transient at $\tau = 0$ dies quickly, and after about 1/3 millisecond the response function closely approximates the input function and indicates the desired harmonic solution is being approached asymptotically. The non-dimensional perturbation pressure is plotted in Figure (4). The complete history is shown for the point $n = 0$, whereas the history of the points $n = 1$ and 2 is only shown at early times, where a difference in amplitude as well as phase exists. The values of p_1 and p_2 are not plotted at later times (that is, after about 1/3 millisecond) since the pressure solutions at the various grid points only differ noticeably in the phase angle corresponding to the input function and the differences in amplitude asymptotically decay. However, the complete numerical results for $n = 0, 1, 2, \dots, 7$ are included in Appendix B as the first test case.

DESCRIPTION OF COMPUTER CODE

Overview

Program RAFFT (Real Acoustic Far-Field Theory), listed in Appendix A, is designed to compute the real part of the acoustic far-field of a pair of interacting cascades in an infinite duct. This program has not as yet been coupled to program B2DATL which computes the near field of the cascades. It currently consists of two parts. The first part, which is made up of the main program RAFFT, the subroutine GETQR (which calculates the input functions Q and R) and the function routine TRØØT, is at present used solely for the test purposes; it must be replaced by a calling routine in B2DATL which calculates the input

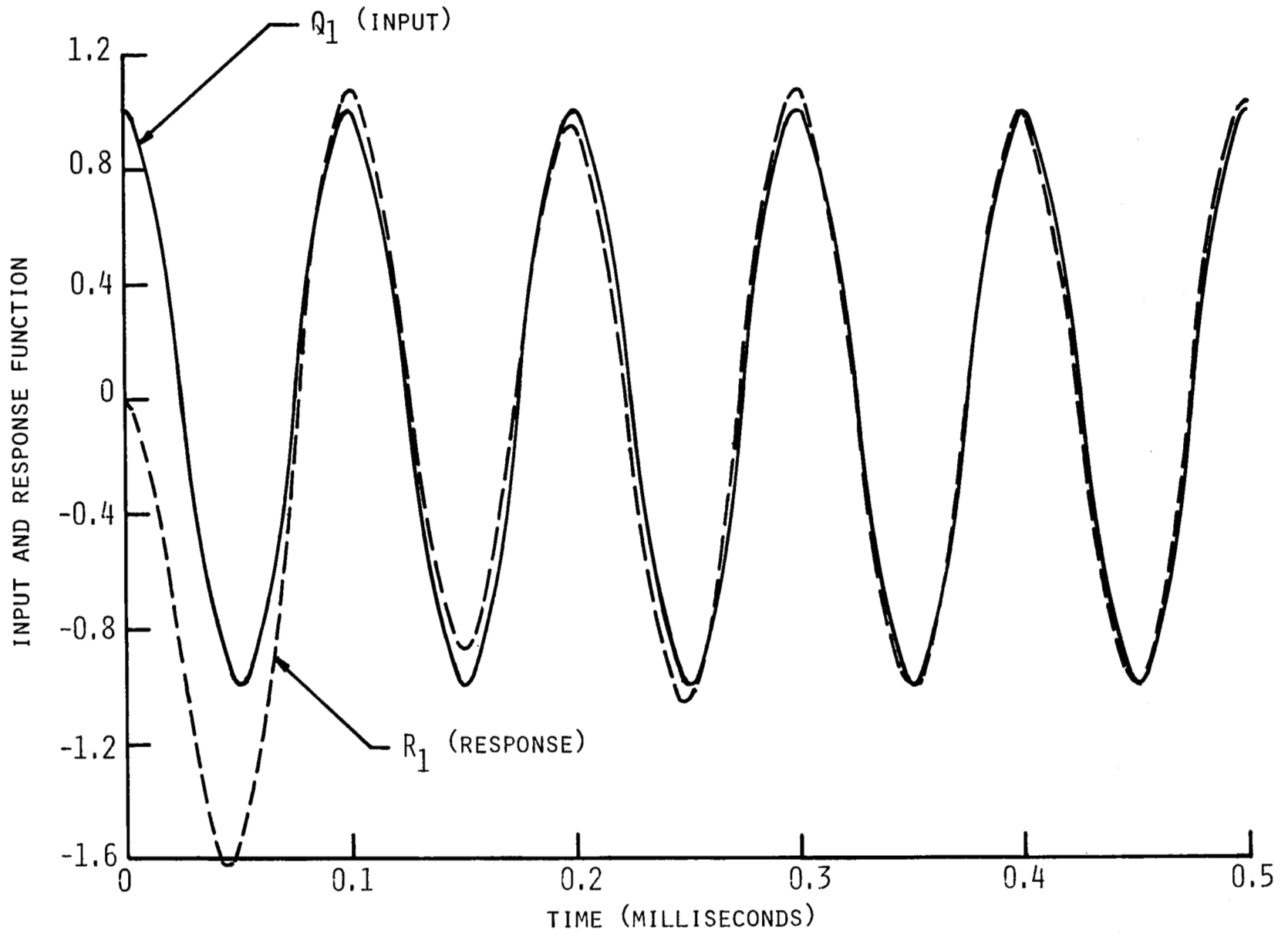


FIGURE 3. INPUT AND RESPONSE FUNCTIONS FOR TEST CASES.

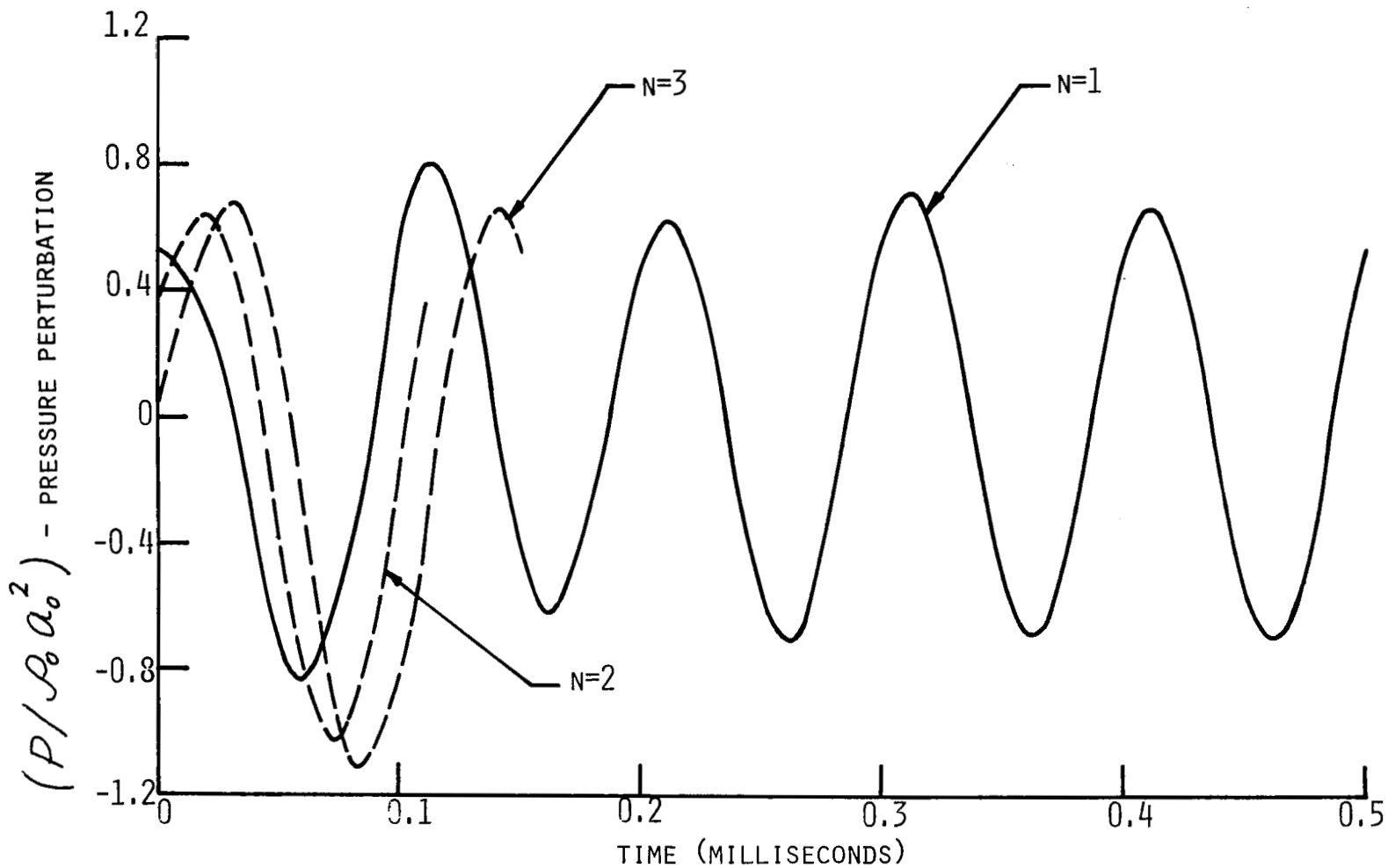


FIGURE 4. PRESSURE PERTURBATIONS AT SEVERAL GRID POINTS

functions Q and R from the boundary data in accord with Equations (38), (39) and (50). The remainder of the program consists of seven subroutines and function routines: AJ1DX, CØNVØL, FFT, FFT2, GREENS, PØLY and TCØNV. Only subroutine CØNVØL is called from the main program; all other subroutines and function routines are called from CØNVØL. Subroutine CØNVØL is called by the statement:

```
CALL CØNVØL (P, Q, R, NY, IT, DTIME, MAXCØN, CMARK, RHØØ, C, Y, FIRST,
            NDIMY, NDIMT, PU)
```

which requires all variables as input parameters except P and PU which are the only results. P(IY, IT) is the discharge station perturbation pressure, as a function of discrete spatial location space, IY and time, IT, and PU(IY,IT) is the corresponding inlet pressure.

Input Data

The input parameters Q(IY,IT), R(IY,IT), NY, IT, DTIME, MAXCØN, CMARK, RHØØ, CY, Y, FIRST, NDIMY and NDIMT, required for the evaluation of P(IY,IT) through the statement CALL CØNVØL (...), are defined as follows:

<u>NAME</u>	<u>DIMENSION</u>	<u>DEFINITION</u>
Q	(8,1001)	Near-field discharge or inlet boundary condition, $(p' + \rho_0 a_0 V_m') / \rho_0 a_0^2$. See Equations (38) and (50).
R	(8,1001)	Circumferential perturbation velocity, V_θ' / a_0 . See Equation (39).
NY		Number of grid points in y-direction.
IT		Current index.
DTIME		Time step; seconds.
MAXCØN		Maximum number of points used for convolution in time (≤ 1001 , with present dimension of Q and R).
CMARK		Reference Mach number, M_0 .
RHØØ		Reference density, ρ_0 ; kg/m ³ .
C		Reference speed of sound, a_0 ; m/sec.

<u>NAME</u>	<u>DIMENSION</u>	<u>DEFINITION</u>
Y		Fundamental wavelength of the cascade ($2\pi r/\Delta N$); m. See Equation (54).
ØMEGAØ		Input frequency (non-dimensional), Ω . See Equation (57).
FIRST		A logical parameter which must be set to "FIRST = .TRUE." before subroutine CØNVØL is called for the first time. This avoids repetition of calculations that need to be done only once. Subroutine CØNVØL automatically sets "FIRST=.FALSE." after the initial call.
NDIMY		Dimension of the arrays Q and R in the Y (circumferential) direction, presently 8. ($NY \leq NDIMY$ is necessary.)
NDIMT		Dimension of the arrays Q and R in time, presently 1001.

Subroutine and Function Sub-Programs

The following seven subroutines and function routines encompass the main portion of the program necessary to determine the acoustic far-field solution:

<u>NAME</u>	<u>DESCRIPTION</u>
CØNVØL	When called the first time, CØNVØL calculates a number of parameters used through the rest of the program, calls subroutine GREENS to calculate the Green's functions H_m and J_m . Thereafter, it calls subroutine TCØNV to perform the convolutions, and finally it evaluates Equations (83):
	$P(IY, IT) = \frac{RHØØ \cdot C \cdot C}{NY} \left(\sum_{KY=1}^{NY} [Q(IY, IT) * H(KY, IT) + R(IY, IT) * J(KY, IT)] \right)$
	where * denotes convolution in time.
AJ1DX	Evaluates the Bessel function $J_1(X)/X$ using a 7th order polynomial approximation.

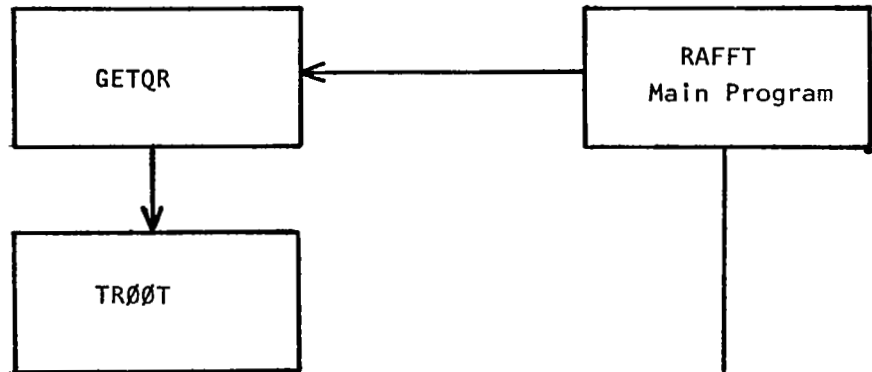
<u>NAME</u>	<u>DESCRIPTION</u>
FFT	Converts a complex array into two real arrays, and then calls FFT2.
FFT2	A fast Fourier transform subroutine written in assembly language*.
GREENS	Calculates the Green's functions H_m , J_m and K_m in accord with Equation (84) for $m = 0, 1, 2, \dots, NY-1$. In this routine the expression $f(W) = W\sqrt{W^2-1}$ occurs where $W = \Omega$ (Equation (57)). The sign of the square root is chosen in such a way that $f(W)$ is symmetric in the imaginary part and antisymmetric in the real part, with $f(0) = +i$. To calculate $H_1(t)$ we evaluate the integral in a finite interval from $-WCUT$ to $+WCUT$ using the fast Fourier transform, where the cut-off frequency is calculated as $WCUT = Y/C/DTIME$. The contribution from the tails outside the interval $(-WCUT, +WCUT)$ is just 1/2 the delta function, plus terms of second order (i.e., containing the factor $1/WCUT^{**2}$).
POLY	Evaluates a polynomial, using as few multiplications as possible.
TCØNV	Forms the convolution of two arrays A and B, calculating only one new value $S_i = \sum_{j=1}^{j_{\max}} A(j) * B(i-j) \quad (\text{for one index } i=1T)$ where $j_{\max} = \min(i, \text{MAXCØN})$, and MAXCØN is the maximum number of points used for convolution.

*Only suitable for use on CDC 6600 computer systems. However, similar FFT routines are generally available for other systems.

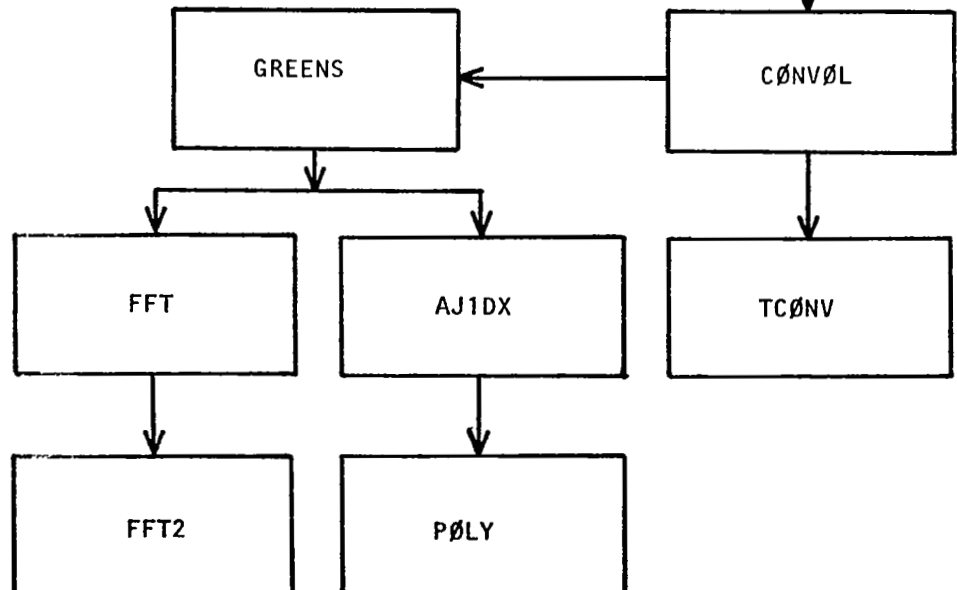
Flow Chart of the Program

The interconnection between the subprograms are represented in the schematic below:

Part 1 (Generates input functions and prints output. Should be replaced by a calling routine from B2DATL.)



Part 2 (Determines acoustic far-field solution)



Test Cases

The input function $Q_n(\tau) = \cos(\Omega \tau - 2\pi \frac{n}{N}) \cdot \Delta(\tau)$, where $\tau = 2\pi N a_0 t / Y$ is non-dimensional time, and $N=NY$ is the number of points in the y -direction, was used for a test example. In this case $R_n(\tau)$ is the response function, rather than input function.

To check $R_n(\tau)$, it was compared with its asymptotic value. Two cases must be distinguished.

1. If $\Omega \geq n\sqrt{1-M^2}$:

$$R_{n, \text{imag}}(\tau) = \frac{1}{n(1-M)} [\Omega - T_n(\Omega)] \sin \Omega \tau \delta_{n-m} \quad (87)$$

$$R_{n, \text{real}}(\tau) = \frac{1}{n(1-M)} [\Omega - T_n(\Omega)] \cos \Omega \tau \delta_{n-m} \quad (88)$$

$$\text{with } T_n(\Omega) = \sqrt{\Omega^2 - n^2 (1-M^2)} \quad (89)$$

2. If $\Omega < n\sqrt{1-M^2}$, substitution of $T_n = -i|T_n|$ yields:

$$R_{n, \text{imag}}(\tau) = \frac{1}{n(1-M)} [\Omega \sin \Omega \tau + T_n \cos \Omega \tau] \delta_{n-m} \quad (90)$$

$$R_{n, \text{real}}(\tau) = \frac{1}{n(1-M)} [\Omega \cos \Omega \tau - T_n \sin \Omega \tau] \delta_{n-m} \quad (91)$$

Two test examples were run, both with $m=1$, and with $\Omega = 1.0$ in the first case and $\Omega = 0.5$ in the second case. The first test case corresponds to the numerical example discussed in Section IV. Due to the factor δ_{n-m} , $R_n = 0$ except for $n=1$.

In the first example $\Omega = 1.0 > n\sqrt{1-M^2} = 0.6$. In the second example $\Omega = 0.5 < n\sqrt{1-M^2} = 0.6$.

As shown in Appendix B, R_1 approaches its asymptotic value reasonably well in both cases.

The print-out in Appendix B includes:

- a list of the values of all parameters used
- a comparison of $R(\tau)$ with its asymptotic value
- the input functions $Q(IY, IT)$ and $R(IY, IT)$
- the Green's functions $H_r(\tau)$ and $J_r(\tau)$ and
- the resulting pressure $P(IY, IT)$.

APPENDIX A
PROGRAM LISTING

```

PROGRAM RAFFT(INPUT,OUTPUT,TAPE98)
DIMENSION P(8,1001),Q(8,1001),R(8,1001),PU(8,1001)
LOGICAL FIRST
DIMENSION ASYMP(8)
COMPLEX EPWNER,TRUOT,E2PI,R1,R7,Ew1,Ew7
COMMON/NSTEP/NSSTEP
EPWNER(X)=CEXP(CMPLX(0,,X))
*
* READ AND PRINT PARAMETERS
10 READ 110,DTIME,NY,NT,MAXCON,CMARK,RHOO,C,Y,WZERO
110 FORMAT(F10.0,3I10,4F10.0)
IF(NT.EQ.0) STOP
PRINT120,DTIME,NY,NT,MAXCON,CMARK,RHOO,C,Y,WZERO
120 FORMAT(*1PARAMETERS USED*/
** DTIME **E10,3/
** NY **I10/
** NT **I10/
** MAXCON **I10/
** CMARK **E10,3/
** RHOO **E10,3/
** C **E10,3/
** Y **E10,3/
** DMEGAO **E10,3/ )
*
NSSTEP=5
NDIMY=8
NDIMT=1001
FIRST=.TRUE.
*
11 CALL SECOND(T0)
PRINT 114
114 FORMAT(/** TEST R1STAR APPROACHES ITS ASYMPTOTIC VALUE*/
DO 12 IT=1,NT
* GET NEW VALUES OF Q AND R AT TIME T FOR NY DISCRETE VALUES OF Y
CALL GETGR(Q,R,IT,DTIME,NY,CMARK,C,Y,NDIMY,NDIMT,WZERO)
12 CONTINUE
*
PRINT 115
115 FORMAT(*1 TIME INPUT R(TIME) *)
DO 15 IT=1,NT,NSSTEP
TIME=(IT-1)*DTIME
15 PRINT 125,TIME,(R(IY,IT),IY=1,NY)
125 FORMAT(1X,F10.3,F9.3,7F7.3)
PRINT 116
116 FORMAT(*1 TIME INPUT Q(TIME)*)
DO 16 IT=1,NT,NSSTEP
TIME=(IT-1)*DTIME
16 PRINT 125,TIME,(Q(IY,IT),IY=1,NY)
CALL SECOND(T1)
TUSED=T1-T0
PRINT 310,TUSED
*
DO 20 IT=1,NT
* CALCULATE P(Y,T) FOR DISCRETE VALUES OF Y
* BY CONVOLUTION WITH GREFNS FUNCTIONS
CALL CONVOL(P,Q,R,NY,IT,DTIME,MAXCON,CMARK,RHOO,C,Y,FIRST,
*
NDIMY,NDIMT,PU)
*
* PRINT SELECTED VALUES OF THE RESULTING PRESSURE P
IF(MOD(IT-1,NSSTEP).NE.0) GO TO 20
IF(IT.EQ.1) PRINT 130
130 FORMAT(*1 TIME PRESSURE P(TIME) FOR DISCRETE VALUES OF Y*)
IF(IT.EQ.1) CALL SECOND(T1)
TIME=(IT-1)*DTIME
PRINT 140,TIME,(P(IY,IT),IY=1,NY)
140 FORMAT(1X,9E10.3)
20 CONTINUE
CALL SECOND(T2)
TUSED=T2-T1
PRINT 310,TUSED
310 FORMAT(/** TIME USED **F7.3* SECONDS**/)
PRINT 150
150 FORMAT(*1 TIME UPSTREAM COMPONENT OF PRESSURE*)
DO 50 IT=1,NT,NSSTEP
TIME=(IT-1)*DTIME
50 PRINT 140,TIME,(PU(IY,IT),IY=1,NY)
END

```

```

SUBROUTINE GETQ(Q,R,IT,DTIME,NY,CMARK,C,Y,NDIMY,NDIMT,WZERO)
DIMENSION Q(NDIMY,NDIMT),R(NDIMY,NDIMT)
REAL J1(1001),J7(1001)
COMPLEX Q1STAR(1001),Q7STAR(1001),R1STAR,R7STAR,RR(8)
COMPLEX ASYMP,TROOT
COMMON/NSTEP/NSTEP
PI=3.1415926535898
TWOPI=2.*PI
I=(IT-1)*DTIME
ALPHAC=TWOPI*C/Y
W0=ALPHAC+WZERO
WOT=W0*T
A=SQRT(1.-CMARK**2)
*
10 DD 10 IY=1,NY
   Q(IY,IT)=COS(WOT-TWOPI*(IY-1)/NY)
   CONTINUE
*
   J1(IT)=(1.+CMARK)*AJ1DX(ALPHAC*A*T)
   Q1STAR(IT)=.5*CEXP(CMPLX(0.,WOT))
   R1STAR=0.
   DO 20 JT=1,IT
   R1STAR=R1STAR+J1(JT)*Q1STAR(IT-JT+1) *(0.,1.) /PI
20 CONTINUE
*
   IF(MOD(IT-1,NSTEP).NE.0) GO TO 25
   ASYMP1=(1./(1.-CMARK)) * (WZERO-TROOT(1,CMARK,WZERO))
*   * CEXP(CMPLX(0.,WOT))
125 PRINT 125,IT,R1STAR,ASYMP1
25 FORMAT(* TIME STEP*14* R1STAR **2F7,3* ASYMPTOTE **2F7,3)
CONTINUE
*
30 DD 30 IY=1,NY
   R(IY,IT)=REAL( CEXP(CMPLX(0.,-TWOPI*(IY-1)/NY))*R1STAR )
CONTINUE
RETURN
END

```

```

COMPLEX FUNCTION TROOT(IY,CMARK,W)
S=W**2-IY**2*(1.-CMARK**2)
IF(S.GE.0.) TROOT=SQRT(S)
IF(S.LT.0.) TROOT=(0.,-1.)*SQRT(-S)
RETURN
END

```



```

SUBROUTINE GREENS(HR,JR,D1,MAXCON,NY,CMARK,P,NDIMP,KR)
*
* GENERATE THE GREENS FUNCTIONS HR(T) AND JR(T)
* FOR I = 0, DT, 2*DT, ..., (MAXCON-1)*DT
* AND FOR R = 0, 1, ..., NY/2 (FOR H), R = 1, 2, ..., NY/2-1 (FOR J)
*
REAL HR(1001,5),JR(1001,3),COSJK(8),SINJK(8)
REAL KR(1001,5)
COMPLEX S,P(1)
COMMON/PI/PI,PIHALF,TWOPI
*
* P, WHICH IS NOT DEFINED AT THE BEGINNING, IS USED
* AS DUMMY STORAGE FOR THE FAST FOURIER TRANSFORM
*
* FIND MAXIMUM N WHICH FITS WITHIN SPACE OF P
N=16384
10 IF(N.LE,NDIMP) GO TO 20
N=N/2
GO TO 10
*
20 CM=1.-CMARK**2
A=SQRT(CM)
WCUT=TWOPI/DT
DN=WCUT/N
NY2=NY/2
NY2I=NY2+1
NYM1=NY-1
NY2M1=NY2-1
N4=N/4
* TABLES OF SINES AND COSINES
DO 25 I=1,NY
ARG=TWOPI*(I-1)/NY
SINJK(I)=SIN(ARG)
25 COSJK(I)=COS(ARG)
*
* CALCULATE H1
1000 FORMAT(1X,F10.3,F9.3,8F7.3)
1180 FORMAT(*1 TIME GREENS FUNCTION HR*)
PRINT 1180
DO 30 I=1,N
W=(I-1)*DN
SQ=W**2-CM
IF(SQ.GE,0.) S=SQRT(SQ)
IF(SQ.LT,0.) S=CMPLX(0.,-SQRT(-SQ))
30 P(I)=(N-S)*W/CM
*
CALL FFT(P,N)
*
DO 40 I=1,N4
* HN IS EVEN, TAKE TWICE REAL PART OF FOURIER TRANSFORM
P(I)=REAL(P(I))*DN/PI
* DELTA FUNCTION
IF(I.EQ,1) P(I)=.5/DT
40 CONTINUE
*
* COSINE SERIES
DO 70 I=1,MAXCON
DO 60 J1=1,NY21
J=J1-1
SUM=0.
* H0 = 1/2 * DELTA FUNCTION
IF(I.EQ,1) SUM=.5/DT
DO 50 K=1,NYM1
IF(K#1 .GT,N4) GO TO 50
INDEX=MOD(J*K,NY)+1
IF(I.GT,1) SUM=SUM+COSJK(INDEX) * K*P(K*I)
IF(I.EQ,1) SUM=SUM+COSJK(INDEX) * P(1)
50 CONTINUE
60 HR(I,J1)=SUM
IF(MOD(I,50).EQ,2 .AND. I.NE,2) PRINT 1180
T=(I-1)*DT
180 PRINT 1000,T,(HR(I,J),J=1,NY21)
70 CONTINUE
*
* JN (FOR N#1) IS THE BESSEL FUNCTION M * J1(A*T)/(A*T)
* WITH A = SQRT(1-M**2)
1190 FORMAT(*1 TIME GREENS FUNCTION JR*)
PRINT 1190
DO 80 I=1,N4
T=(I-1)*DT
80 P(I)=CMARK*AJ1DX(A*T)

```

```

*
* SINE SERIES
  DO 190 I=1,MAXCON
  DO 100 J=1,NY2M1
  SUM=0.
  DO 90 K=1,NYM1
  IF(K*I .GT.N4) GO TO 90
  INDEX=MOD(J*K,NY)+1
  SUM=SUM+SINJK(INDEX) * K*P(K*1 )
90  CONTINUE
100  JR(I,J)=SUM
  IF(MOD(I,50).EQ.2 .AND. I.NE.2) PRINT 1190
  T=(I-1)*DT
  PRINT 1000,T,(JR(I,J),J=1,NY2M1)
190  CONTINUE
*
* CALCULATE K1
1200 FORMAT(*1          TIME      GREENS FUNCTION KR*)
  PRINT 1200
  DO 130 I=1,N
  W=(I-1)*DW
  SQ=W**2-CM
  IF(SQ.GE.0.) S=SQRT(SQ)
  IF(SQ.LT.0.) S=CMPLX(0.,-SQRT(-SQ))
130  P(I)=(N-S)*(K+CMARK*S)/(CM*(1.+CMARK))
  CALL FFI(P,N)
  DO 140 I=1,N4
  P(I)=REAL(P(I))*DW/PI
  IF(I.EQ.1) P(I)=.5/DT
140  CONTINUE
  DO 170 I=1,MAXCON
  DO 160 J1=1,NY21
  J=J1-1
  SUM=0.
  IF(I.EQ.1) SUM=.5/DT
  DO 150 K=1,NYM1
  IF(K*I .GT.N4) GO TO 150
  INDEX=MOD(J*K,NY)+1
  IF(I.GT.1) SUM=SUM+COSJK(INDEX)*K*P(K*1 )
  IF(I.EQ.1) SUM=SUM+COSJK(INDEX)*P(1)
150  CONTINUE
160  KR(I,J1)=SUM
  IF(MOD(I,50).EQ.2 .AND. I.NE.2) PRINT 1200
  T=(I-1)*DT
  PRINT 1000,T,(KR(I,J),J=1,NY21)
170  CONTINUE
*
  RETURN
  END

```



```

      SUBROUTINE TCONV(QR,HJ,IT,DT,NY,MAXCON,RESULT)
*
* CONVOLUTION IN TIME
* OF Q(IY,IT) WITH H(JY,IT)
* OR R(IY,IT) WITH J(JY,IT)
* TO CALCULATE ONE NEW POINT
*
      DIMENSION QR(1),HJ(1)
      RESULT=0.
      MAXT=MAXCON
      IF(IT.LT,MAXCON) MAXT=IT
      J=(IT-1)*NY+1
      DO 10 K=1,MAXT
      RESULT=RESULT+QR(J)*HJ(K)
10    J=J-NY
      RESULT=RESULT*DT
      RETURN
      END

      FUNCTION AJ1DX(X)
* BESSEL FUNCTION J1(X)/X
      DIMENSION C1(7),C2(7),C3(7)
      DATA (C1(I),I=1;7)/.5,=.56249085,.21093573,=.03954289,
*          .00443319,=.00031761,.00001109/
      DATA (C2(I),I=1;7)/.79788456,.00000156,.01658667,.00017105,
*          .00249511,.00113653,.00020033/
      DATA (C3(I),I=1;7)/2.35619449,.12499612,.00065650,=.00637879,
*          .00074348,.00079824,=.00029166/
      AX=ABS(X)
      IF(AX.GT.3.) GO TO 10
      AJ1DX=POLY(C1,(X/3.))**2,7)
      RETURN
10    F1=POLY(C2,3./AX,7)
      THETA1=AX+POLY(C3,3./AX,7)
      AJ1DX=F1*COS(THETA1)/SQRT(AX)/AX
      RETURN
      END

      FUNCTION POLY(A,X,N)
* EVALUATE THE POLYNOMIAL A(1)+A(2)*X+...+A(N)*X**(N-1)
      DIMENSION A(1)
      POLY=0.
      XPOWER=1.
      DO 10 I=1,N
      POLY=POLY+A(I)*XPOWER
10    XPOWER=XPOWER*X
      RETURN
      END

      SUBROUTINE FFT(XREAL,NSIG)
      REAL XREAL(1)
      IF(NSIG.LT.0) GO TO 10
      N=NSIG
      NSPACE=2
      GO TO 20
10    N=-NSIG
      NSPACE=-2
20    CALL FFT2(XREAL,XREAL(2),N,NSPACE)
      RETURN
      END

```

	IDENT	FFT2			NX5	X4
	ENTRY	FFT2			RX7	X1-X6
L100	SX0	B3			RX0	X0+X5
	SB4	B0		L30	NX1	X7
	SB3	B3-B7			SB5	B6+B4
	AX0	1			SA2	B1+B4
	SB5	B0			SA3	B1+B5
	SB6	X0			SA4	B2+B4
	SX1	B5			RX6	X2+X3
	EQ	B6,B7,FFT23			SA5	B2+B5
L110	SB4	B3-B4			RX2	X2-X3
	SB5	B3-B5			SA6	A2
	SA2	B1+B4			RX7	X4+X5
	SA3	B1+B5			RX3	X1+X2
	SA4	B2+B4			RX4	X4-X5
	NX7	X2			SA7	A4
	SA5	B2+B5			RX5	X0+X4
	NX6	X3			RX2	X0+X2
	SA7	A3			RX6	X3-X5
	SA6	A2			RX4	X1+X4
	NX7	X4			SA6	A3
	NX6	X5			RX7	X2+X4
	SA7	A5			SB4	B6+B5
	SA6	A4			SA7	A5
	LT	B6,B4,L110			LT	B4,B3,L30
L120	SB4	B4+B7			SB5	B4-B3
	SB5	B6+B5			BX1	-X1
	SA2	B1+B4			SB4	B6-B5
	SA3	B1+B5			LT	B5,B4,L30
	SA4	B2+B4			SB4	B4+B7
	NX7	X2			SA2	SD
	SA5	B2+B5			LT	B4,B5,L20
	NX6	X3		L40	SB4	B0
	SA7	A3			SX5	B6
	SA6	A2			AX5	1
	NX7	X4			SB6	X5
	SX0	B6		L50	SB5	B6+B4
	NX6	X5			SA2	B1+B4
	SA7	A5			SA3	B1+B5
	SA6	A4			SA4	B2+B4
L130	AX0	1			RX6	X2+X3
	IX1	X1-X0			SA5	B2+B5
	PL	X1,L130			RX7	X2-X3
	LX0	1			SA6	A2
	SB4	B4+B7			SA7	A3
	IX1	X1+X0			RX6	X4+X5
	SB5	X1			SB4	B6+B5
	GE	B5,B4,L110			RX7	X4-X5
	LT	B4,B6,L120			SA6	A4
FFT23	SA1	A0			SA7	A5
FFT2					LT	B4,B3,L50
	SA0	A1			EQ	B6,B7,L100
	SA1	A0			SA1	A1
	SB1	X1			SB4	B7
	SA1	A0+1			RX6	X1+X1
	SB2	X1			SA1	A1+1
	SA1	A0+2			FX6	X6+X6
	SB3	X1			SA3	ONE
	SA1	A0+3			SA6	CD
	SB4	X1		L60	BX0	X1
	SA4	B4			RX6	X3-X6
	MX2	1			BX7	X0
	SA5	L60			NX1	X6
	SA3	B3			SA7	SD
	LX2	57			EQ	L30
	PX7	X3		S	DATA	9.5873799095977346E-5
	BX6	-X2+X5			DATA	1.9174759731070331E-4
	PL	X4,L10			DATA	3.834951875713959E-4
	BX6	X2+X5			DATA	7.6699031874270453E-4
	BX4	-X4			DATA	1.5335801862847656E-3
L10	LX3	32			DATA	3.0679567629659763E-3
	SA6	A5			DATA	6.1358846491544754E-3
	NX0	B5,X3			DATA	1.2271538285719926E-2
	PX2	X4			DATA	2.454122852291228E-2
	SB7	X4			DATA	4.9067674327418014E-2
	DX7	X2+X7			DATA	9.8017140329560602E-2
	SA1	B5+S			DATA	1.9509032201612427E-1
	SB3	X7			DATA	3.8268343236500977E-1
	SB6	X7			DATA	0.7071067811865475
	EQ	L40		ONE	DATA	1.0
L20	SA3	CD		CD		
	RX4	X2+X1		SD		
	RX7	X2+X0				
	RX5	X3+X0				
	RX6	X3+X1				
	RX4	X4-X5				
	RX6	X6+X7				

END

APPENDIX B
PRINTOUT OF TEST CASES

PARAMETERS USED
 DTIME = .250E-05
 NY = 8
 NT = 201
 MAXCON = 101
 CMARK = .800E+00
 RHOO = .100E+01
 C = .100E+04
 Y = .100E+00
 OMEGA0 = .100E+01

TEST R1STAR APPROACHES ITS ASYMPTOTIC VALUE

TIME STEP	1	R1STAR =	0.000	.143	ASYMPTOTE =	1.000	0.000
TIME STEP	6	R1STAR =	-.316	.757	ASYMPTOTE =	.707	.707
TIME STEP	11	R1STAR =	-.963	.928	ASYMPTOTE =	.000	1.000
TIME STEP	16	R1STAR =	-1.520	.520	ASYMPTOTE =	-.707	.707
TIME STEP	21	R1STAR =	-1.596	-.257	ASYMPTOTE =	-1.000	.000
TIME STEP	26	R1STAR =	-1.065	-.969	ASYMPTOTE =	-.707	-.707
TIME STEP	31	R1STAR =	-.150	-1.209	ASYMPTOTE =	-.000	-1.000
TIME STEP	36	R1STAR =	.701	-.835	ASYMPTOTE =	.707	-.707
TIME STEP	41	R1STAR =	1.068	-.057	ASYMPTOTE =	1.000	-.000
TIME STEP	46	R1STAR =	.801	.687	ASYMPTOTE =	.707	.707
TIME STEP	51	R1STAR =	.101	.983	ASYMPTOTE =	.000	1.000
TIME STEP	56	R1STAR =	-.596	.680	ASYMPTOTE =	-.707	.707
TIME STEP	61	R1STAR =	-.874	-.023	ASYMPTOTE =	-1.000	.000
TIME STEP	66	R1STAR =	-.580	-.698	ASYMPTOTE =	-.707	-.707
TIME STEP	71	R1STAR =	.098	-.940	ASYMPTOTE =	-.000	-1.000
TIME STEP	76	R1STAR =	.738	-.603	ASYMPTOTE =	.707	-.707
TIME STEP	81	R1STAR =	.944	.113	ASYMPTOTE =	1.000	-.000
TIME STEP	86	R1STAR =	.577	.782	ASYMPTOTE =	.707	.707
TIME STEP	91	R1STAR =	-.159	1.004	ASYMPTOTE =	.000	1.000
TIME STEP	96	R1STAR =	-.835	.638	ASYMPTOTE =	-.707	.707
TIME STEP	101	R1STAR =	-1.052	-.110	ASYMPTOTE =	-1.000	.000
TIME STEP	106	R1STAR =	-.672	-.806	ASYMPTOTE =	-.707	-.707
TIME STEP	111	R1STAR =	.094	-1.046	ASYMPTOTE =	-.000	-1.000
TIME STEP	116	R1STAR =	.810	-.687	ASYMPTOTE =	.707	-.707
TIME STEP	121	R1STAR =	1.066	.063	ASYMPTOTE =	1.000	-.000
TIME STEP	126	R1STAR =	.717	.772	ASYMPTOTE =	.707	.707
TIME STEP	131	R1STAR =	-.031	1.029	ASYMPTOTE =	.000	1.000
TIME STEP	136	R1STAR =	-.743	.689	ASYMPTOTE =	-.707	.707
TIME STEP	141	R1STAR =	-1.010	-.045	ASYMPTOTE =	-1.000	.000
TIME STEP	146	R1STAR =	-.683	-.743	ASYMPTOTE =	-.707	-.707
TIME STEP	151	R1STAR =	.039	-.996	ASYMPTOTE =	-.000	-1.000
TIME STEP	156	R1STAR =	.726	-.660	ASYMPTOTE =	.707	-.707
TIME STEP	161	R1STAR =	.974	.065	ASYMPTOTE =	1.000	-.000
TIME STEP	166	R1STAR =	.637	.751	ASYMPTOTE =	.707	.707
TIME STEP	171	R1STAR =	-.084	.991	ASYMPTOTE =	.000	1.000
TIME STEP	176	R1STAR =	-.762	.644	ASYMPTOTE =	-.707	.707
TIME STEP	181	R1STAR =	-.993	-.088	ASYMPTOTE =	-1.000	.000
TIME STEP	186	R1STAR =	-.637	-.775	ASYMPTOTE =	-.707	-.707
TIME STEP	191	R1STAR =	.102	-1.011	ASYMPTOTE =	-.000	-1.000
TIME STEP	196	R1STAR =	.791	-.657	ASYMPTOTE =	.707	-.707
TIME STEP	201	R1STAR =	1.028	.085	ASYMPTOTE =	1.000	-.000

TIME	INPUT R(TIME)							
0.	0.000	.101	.143	.101	.000	-.101	-.143	-.101
.125E-04	-.316	.312	.757	.759	.316	-.312	-.757	-.759
.250E-04	-.963	-.025	.928	1.337	.963	.025	-.928	-1.337
.375E-04	-1.520	-.707	.520	1.442	1.520	.707	-.520	-1.442
.500E-04	-1.596	-1.310	-.257	.946	1.596	1.310	.257	-.946
.625E-04	-1.065	-1.439	-.969	.068	1.065	1.439	.969	-.068
.750E-04	-.150	-.961	-1.209	-.749	.150	.961	1.209	.749
.875E-04	.701	-.095	-.835	-1.066	-.701	.095	.835	1.066
.100E-03	1.068	.715	-.057	-.796	-1.068	-.715	.057	.796
.113E-03	.801	1.052	.687	-.081	-.801	-1.052	-.687	.081
.125E-03	.101	.766	.983	.623	-.101	-.766	-.983	-.623
.138E-03	-.596	.060	.680	.902	.596	-.060	-.680	-.902
.150E-03	-.874	-.634	-.023	.602	.874	.634	.023	-.602
.163E-03	-.580	-.904	-.698	-.084	.580	.904	.698	.084
.175E-03	.098	-.596	-.940	-.734	-.098	.596	.940	.734
.188E-03	.738	.095	-.603	-.948	-.738	-.095	.603	.948
.200E-03	.944	.748	.113	-.588	-.944	-.748	-.113	.588
.213E-03	.577	.962	.782	.145	-.577	-.962	-.782	-.145
.225E-03	-.159	.598	1.004	.822	.159	-.598	-1.004	-.822
.238E-03	-.835	-.140	.638	1.041	.835	.140	-.638	-1.041
.250E-03	-1.052	-.822	-.110	.666	1.052	.822	.110	-.666
.263E-03	-.672	-1.046	-.806	-.095	.672	1.046	.806	.095
.275E-03	.094	-.673	-1.046	-.806	.673	1.046	.806	.095
.288E-03	.810	.086	-.687	-1.058	-.810	-.086	.687	1.058
.300E-03	1.066	.798	.063	-.709	-1.066	-.798	-.063	.709
.313E-03	.717	1.053	.772	.039	-.717	-1.053	-.772	-.039
.325E-03	-.031	.706	1.029	.749	.031	-.706	-1.029	-.749
.338E-03	-.743	-.038	.689	1.013	.743	.038	-.689	-1.013
.350E-03	-1.010	-.746	-.045	.683	1.010	.746	.045	-.683
.363E-03	-.683	-1.008	-.743	-.042	.683	1.008	.743	.042
.375E-03	.039	-.677	-.996	-.732	-.039	.677	.996	.732
.388E-03	.726	.047	-.660	-.980	-.726	-.047	.660	.980
.400E-03	.974	.735	.065	-.643	-.974	-.735	-.065	.643
.413E-03	.637	.981	.751	.080	-.637	-.981	-.751	-.080
.425E-03	-.084	.642	.991	.760	.084	-.642	-.991	-.760
.438E-03	-.762	-.063	.644	.994	.762	.063	-.644	-.994
.450E-03	-.993	-.764	-.088	.640	.993	.764	.088	-.640
.463E-03	-.637	-.998	-.775	-.097	.637	.998	.775	.097
.475E-03	.102	-.643	-1.011	-.787	-.102	.643	1.011	.787
.488E-03	.791	.095	-.657	-1.024	-.791	-.095	.657	1.024
.500E-03	1.028	.787	.085	-.667	-1.028	-.787	-.085	.667

TIME	INPUT Q(TIME)							
0.	1.000	.707	-.000	-.707	-1.000	-.707	-.000	.707
.125E-04	.707	1.000	.707	.000	-.707	-1.000	-.707	-.000
.250E-04	.000	.707	1.000	.707	-.000	-.707	-1.000	-.707
.375E-04	-.707	.000	.707	1.000	.707	-.000	-.707	-1.000
.500E-04	-1.000	-.707	.000	.707	1.000	.707	.000	-.707
.625E-04	-.707	-1.000	-.707	.000	.707	1.000	.707	.000
.750E-04	-.000	-.707	-1.000	-.707	.000	.707	1.000	.707
.875E-04	.707	-.000	-.707	-1.000	-.707	.000	.707	1.000
.100E-03	1.000	.707	-.000	-.707	-1.000	-.707	.000	.707
.113E-03	.707	1.000	.707	-.000	-.707	-1.000	-.707	.000
.125E-03	.000	.707	1.000	.707	-.000	-.707	-1.000	-.707
.138E-03	-.707	.000	.707	1.000	.707	-.000	-.707	-1.000
.150E-03	-1.000	-.707	.000	.707	1.000	.707	.000	-.707
.163E-03	-.707	-1.000	-.707	.000	.707	1.000	.707	-.000
.175E-03	-.000	-.707	-1.000	-.707	.000	.707	1.000	.707
.188E-03	.707	-.000	-.707	-1.000	-.707	.000	.707	1.000
.200E-03	1.000	.707	-.000	-.707	-1.000	-.707	.000	.707
.213E-03	.707	1.000	.707	-.000	-.707	-1.000	-.707	.000
.225E-03	.000	.707	1.000	.707	-.000	-.707	-1.000	-.707
.238E-03	-.707	.000	.707	1.000	.707	-.000	-.707	-1.000
.250E-03	-1.000	-.707	.000	.707	1.000	.707	.000	-.707
.263E-03	-.707	-1.000	-.707	.000	.707	1.000	.707	-.000
.275E-03	-.000	-.707	-1.000	-.707	.000	.707	1.000	.707
.288E-03	.707	-.000	-.707	-1.000	-.707	.000	.707	1.000
.300E-03	1.000	.707	-.000	-.707	-1.000	-.707	.000	.707
.313E-03	.707	1.000	.707	-.000	-.707	-1.000	-.707	.000
.325E-03	.000	.707	1.000	.707	-.000	-.707	-1.000	-.707
.338E-03	-.707	.000	.707	1.000	.707	-.000	-.707	-1.000
.350E-03	-1.000	-.707	.000	.707	1.000	.707	.000	-.707
.363E-03	-.707	-1.000	-.707	.000	.707	1.000	.707	-.000
.375E-03	-.000	-.707	-1.000	-.707	.000	.707	1.000	.707
.388E-03	.707	-.000	-.707	-1.000	-.707	.000	.707	1.000
.400E-03	1.000	.707	-.000	-.707	-1.000	-.707	.000	.707
.413E-03	.707	1.000	.707	-.000	-.707	-1.000	-.707	.000
.425E-03	.000	.707	1.000	.707	-.000	-.707	-1.000	-.707
.438E-03	-.707	.000	.707	1.000	.707	-.000	-.707	-1.000
.450E-03	-1.000	-.707	.000	.707	1.000	.707	.000	-.707
.463E-03	-.707	-1.000	-.707	.000	.707	1.000	.707	-.000
.475E-03	-.000	-.707	-1.000	-.707	.000	.707	1.000	.707
.488E-03	.707	-.000	-.707	-1.000	-.707	.000	.707	1.000
.500E-03	1.000	.707	-.000	-.707	-1.000	-.707	.000	.707

TIME	GREENS	FUNCTION	HR		
0,000	25,465	-.000	-.000	-.000	0,000
.157	-1,681	.078	.284	.317	.324
.314	-2,305	.183	.370	.398	.404
.471	-2,630	.332	.389	.395	.397
.628	-2,648	.508	.341	.318	.314
.785	-2,408	.678	.240	.194	.185
.942	-1,999	.804	.111	.060	.050
1,100	-1,525	.855	-.016	-.049	-.055
1,257	-1,081	.815	-.113	-.108	-.107
1,414	-.731	.688	-.159	-.111	-.104
1,571	-.505	.497	-.146	-.069	-.058
1,728	-.395	.280	-.077	-.007	.003
1,885	-.367	.076	.030	.051	.054
2,042	-.379	-.083	.153	.082	.076
2,199	-.392	-.180	.267	.077	.064
2,356	-.384	-.213	.351	.040	.028
2,513	-.348	-.197	.390	-.011	-.016
2,670	-.293	-.155	.382	-.056	-.047
2,827	-.236	-.110	.332	-.078	-.054
2,985	-.191	-.078	.255	-.064	-.034
3,142	-.167	-.067	.167	-.018	.003
3,299	-.162	-.075	.085	.052	.039
3,456	-.170	-.091	.020	.126	.059
3,613	-.179	-.106	-.020	.189	.053
3,770	-.181	-.111	-.037	.228	.021
3,927	-.172	-.103	-.036	.237	-.023
4,084	-.153	-.087	-.025	.219	-.061
4,241	-.132	-.067	-.013	.183	-.075
4,398	-.114	-.051	-.006	.139	-.051
4,555	-.104	-.044	-.006	.097	.011
4,712	-.104	-.045	-.014	.061	.100
4,869	-.110	-.052	-.026	.034	.197
5,027	-.117	-.060	-.037	.015	.280
5,184	-.120	-.064	-.043	.002	.329
5,341	-.116	-.060	-.043	-.007	.336
5,498	-.106	-.051	-.036	-.011	.303
5,655	-.093	-.040	-.027	-.008	.242
5,812	-.083	-.030	-.017	.004	.170
5,969	-.078	-.025	-.012	.024	.104
6,126	-.079	-.026	-.012	.050	.056
6,283	-.085	-.032	-.018	.077	.031
6,440	-.091	-.038	-.028	.099	.025
6,597	-.094	-.042	-.037	.111	.030
6,754	-.092	-.041	-.042	.110	.037
6,912	-.084	-.034	-.039	.095	.039
7,069	-.075	-.025	-.027	.072	.034
7,226	-.067	-.016	-.008	.046	.022
7,383	-.064	-.011	.017	.022	.008
7,540	-.066	-.013	.041	.006	-.004
7,697	-.072	-.020	.061	-.001	-.010
7,854	-.079	-.029	.073	-.000	-.009

TIME	GREENS	FUNCTION	HR		
8,011	-.083	-.036	.075	.004	-.005
8,168	-.081	-.037	.069	.009	.000
8,325	-.074	-.030	.055	.010	.003
8,482	-.064	-.014	.038	.006	.002
8,639	-.055	.006	.021	.002	-.003
8,796	-.052	.027	.009	-.005	-.009
8,954	-.057	.045	.001	-.011	-.014
9,111	-.067	.056	-.001	-.013	-.016
9,268	-.079	.059	.000	-.012	-.015
9,425	-.087	.055	.003	-.008	-.011
9,582	-.085	.046	.005	-.005	-.008
9,739	-.071	.035	.006	-.003	-.005
9,896	-.044	.024	.004	-.003	-.005
10,053	-.009	.015	-.000	-.006	-.008
10,210	.025	.009	-.005	-.010	-.012
10,367	.053	.003	-.008	-.014	-.015
10,524	.067	-.000	-.010	-.015	-.016
10,681	.066	-.003	-.010	-.014	-.015
10,838	.052	-.003	-.007	-.010	-.012
10,996	.029	.000	-.004	-.007	-.008
11,153	.005	.007	-.002	-.005	-.006
11,310	-.015	.017	-.001	-.005	-.006
11,467	-.011	.041	-.003	-.020	-.026
11,624	-.021	.048	-.007	-.019	-.024
11,781	-.026	.052	-.011	-.018	-.020
11,938	-.026	.053	-.015	-.016	-.017
12,095	-.024	.050	-.015	-.015	-.016
12,252	-.021	.045	-.012	-.014	-.016
12,409	-.020	.038	-.005	-.014	-.018
12,566	-.021	.031	.005	-.015	-.021
12,723	-.024	.024	.016	-.017	-.024
12,881	-.029	.019	.027	-.019	-.025
13,038	-.034	.015	.034	-.020	-.025
13,195	-.037	.013	.038	-.021	-.023
13,352	-.023	.013	.021	-.020	-.004
13,509	-.027	.013	.020	-.016	-.003
13,666	-.030	.013	.017	-.013	-.005
13,823	-.032	.013	.013	-.007	-.008
13,980	-.033	.013	.008	.001	-.012
14,137	-.032	.012	.003	.000	-.016
14,294	-.031	.011	-.001	.015	-.019
14,451	-.029	.009	-.005	.020	-.020
14,608	-.028	.008	-.008	.022	-.018
14,765	-.027	.007	-.009	.022	-.012
14,923	-.027	.006	-.010	.019	-.003
15,080	-.028	.006	-.009	.014	.007
15,237	-.030	.006	-.008	.007	.020
15,394	-.032	.007	-.007	.001	.031
15,551	-.034	.007	-.006	-.005	.042
15,708	-.035	.007	-.006	-.009	.049

TIME	GREENS	FUNCTION	JR
0,000	-3,698	-1,520	-,629
,157	-3,114	-1,248	-,514
,514	-2,208	-,851	-,348
,471	-1,138	-,420	-,171
,628	-,078	-,054	-,024
,785	,620	,185	,072
,942	1,448	,264	,098
1,100	1,764	,198	,073
1,257	1,788	,050	,020
1,414	1,590	-,108	-,032
1,571	1,267	-,208	-,059
1,728	,912	-,203	-,055
1,885	,601	-,090	-,024
2,042	,375	,105	,017
2,199	,242	,329	,046
2,356	,184	,525	,050
2,513	,172	,650	,024
2,670	,174	,681	-,020
2,827	,170	,625	-,062
2,985	,152	,510	-,080
3,142	,123	,373	-,056
3,299	,091	,249	,012
3,456	,065	,163	,114
3,613	,051	,121	,228
3,770	,048	,113	,325
3,927	,052	,122	,383
4,084	,055	,131	,390
4,241	,054	,128	,348
4,398	,048	,111	,273
4,555	,037	,085	,188
4,712	,027	,059	,117
4,869	,020	,043	,075
5,027	,019	,039	,066
5,184	,021	,046	,082
5,341	,025	,055	,108
5,498	,027	,061	,125
5,655	,026	,058	,120
5,812	,021	,045	,089
5,969	,014	,029	,036
6,126	,009	,016	-,027
6,283	,007	,011	-,086
6,440	,009	,017	-,127
6,597	,013	,031	-,145
6,754	,018	,045	-,139
6,912	,019	,051	-,116
7,069	,015	,042	-,086
7,226	,009	,016	-,057
7,383	,003	-,023	-,037
7,540	-,001	-,066	-,028
7,697	,001	-,103	-,027
7,854	,008	-,125	-,030

TIME	GREENS	FUNCTION	JR
8,011	,018	-,129	-,033
8,168	,024	-,116	-,033
8,325	,023	-,093	-,029
8,482	,011	-,066	-,023
8,639	-,013	-,044	-,017
8,796	-,045	-,032	-,013
8,954	-,077	-,029	-,012
9,111	-,101	-,032	-,014
9,268	-,113	-,037	-,016
9,425	-,109	-,039	-,016
9,582	-,093	-,037	-,016
9,739	-,069	-,030	-,013
9,896	-,046	-,022	-,010
10,053	-,029	-,016	-,008
10,210	-,023	-,014	-,007
10,367	-,027	-,016	-,007
10,524	-,037	-,020	-,009
10,681	-,046	-,023	-,010
10,838	-,049	-,024	-,011
10,996	-,041	-,020	-,009
11,153	-,024	-,015	-,007
11,310	,000	-,009	-,005
11,467	,019	-,015	-,010
11,624	,037	-,020	-,013
11,781	,050	-,023	-,013
11,938	,057	-,023	-,011
12,095	,056	-,020	-,006
12,252	,048	-,012	-,001
12,409	,037	-,002	,002
12,566	,025	,011	,003
12,723	,015	,024	,001
12,881	,008	,036	-,004
13,038	,006	,045	-,010
13,195	,007	,050	-,015
13,352	,004	,051	-,010
13,509	,004	,048	-,003
13,666	,005	,042	,006
13,823	,008	,034	,016
13,980	,011	,026	,025
14,137	,013	,019	,033
14,294	,014	,014	,039
14,451	,013	,011	,042
14,608	,011	,011	,042
14,765	,008	,012	,041
14,923	,005	,014	,039
15,080	,003	,016	,035
15,237	,001	,017	,032
15,394	,001	,016	,028
15,551	,001	,015	,025
15,708	,003	,012	,022

TIME	GREENS	FUNCTION	KR		
0.000	25.465	-.000	-.000	-.000	0.000
.157	-.187	.009	.032	.035	.036
.314	-.256	.020	.041	.044	.045
.471	-.292	.037	.043	.044	.044
.628	-.294	.056	.038	.035	.035
.785	-.268	.075	.027	.022	.021
.942	-.222	.089	.012	.007	.006
1.100	-.169	.095	-.002	-.005	-.006
1.257	-.120	.091	-.013	-.012	-.012
1.414	-.081	.076	-.018	-.012	-.012
1.571	-.056	.055	-.016	-.008	-.006
1.728	-.044	.031	-.009	-.001	.000
1.885	-.041	.008	.003	.006	.006
2.042	-.042	-.009	.017	.009	.008
2.199	-.044	-.020	.030	.009	.007
2.356	-.043	-.024	.039	.004	.003
2.513	-.039	-.022	.043	-.001	-.002
2.670	-.033	-.017	.042	-.006	-.005
2.827	-.026	-.012	.037	-.009	-.006
2.985	-.021	-.009	.028	-.007	-.004
3.142	-.019	-.007	.019	-.002	.000
3.299	-.018	-.008	.009	.006	.004
3.456	-.019	-.010	.002	.014	.007
3.613	-.020	-.012	-.002	.021	.006
3.770	-.020	-.012	-.004	.025	.002
3.927	-.019	-.011	-.004	.026	-.003
4.084	-.017	-.010	-.003	.024	-.007
4.241	-.015	-.007	-.001	.020	-.008
4.398	-.013	-.006	-.001	.015	-.006
4.555	-.012	-.005	-.001	.011	.001
4.712	-.012	-.005	-.002	.007	.011
4.869	-.012	-.006	-.003	.004	.022
5.027	-.013	-.007	-.004	.002	.031
5.184	-.013	-.007	-.005	.000	.037
5.341	-.013	-.007	-.005	-.001	.037
5.498	-.012	-.006	-.004	-.001	.034
5.655	-.010	-.004	-.003	-.001	.027
5.812	-.009	-.003	-.002	.000	.019
5.969	-.009	-.003	-.001	.003	.012
6.126	-.009	-.003	-.001	.006	.006
6.283	-.009	-.004	-.002	.009	.003
6.440	-.010	-.004	-.003	.011	.003
6.597	-.010	-.005	-.004	.012	.003
6.754	-.010	-.005	-.005	.012	.004
6.912	-.009	-.004	-.004	.011	.004
7.069	-.006	-.003	-.003	.008	.004
7.226	-.007	-.002	-.001	.005	.002
7.383	-.007	-.001	.002	.002	.001
7.540	-.007	-.001	.005	.001	-.000
7.697	-.008	-.002	.007	-.000	-.001
7.854	-.009	-.003	.008	-.000	-.001

TIME	GREENS	FUNCTION	KR		
8.011	-.009	-.004	.008	.000	-.001
8.168	-.009	-.004	.008	.001	.000
8.325	-.008	-.003	.006	.001	.000
8.482	-.007	-.002	.004	.001	.000
8.639	-.006	.001	.002	.000	-.000
8.796	-.006	.003	.001	-.001	-.001
8.954	-.006	.005	.000	-.001	-.002
9.111	-.007	.006	-.000	-.001	-.002
9.268	-.009	.007	.000	-.001	-.002
9.425	-.010	.006	.000	-.001	-.001
9.582	-.009	.005	.001	-.001	-.001
9.739	-.008	.004	.001	-.000	-.001
9.896	-.005	.003	.000	-.000	-.001
10.053	-.001	.002	-.000	-.001	-.001
10.210	.003	.001	-.001	-.001	-.001
10.367	.006	.000	-.001	-.002	-.002
10.524	.007	-.000	-.001	-.002	-.002
10.681	.007	-.000	-.001	-.002	-.002
10.838	.006	-.000	-.001	-.001	-.001
10.996	.003	.000	-.000	-.001	-.001
11.153	.001	.001	-.000	-.001	-.001
11.310	-.002	.002	-.000	-.001	-.001
11.467	-.001	.005	-.000	-.002	-.003
11.624	-.002	.005	-.001	-.002	-.003
11.781	-.003	.006	-.001	-.002	-.002
11.938	-.003	.006	-.002	-.002	-.002
12.095	-.003	.006	-.002	-.002	-.002
12.252	-.002	.005	-.001	-.002	-.002
12.409	-.002	.004	-.001	-.002	-.002
12.566	-.002	.003	.001	-.002	-.002
12.723	-.003	.003	.002	-.002	-.003
12.881	-.003	.002	.003	-.002	-.003
13.038	-.004	.002	.004	-.002	-.003
13.195	-.004	.001	.004	-.002	-.003
13.352	-.003	.001	.002	-.002	-.000
13.509	-.003	.001	.002	-.002	-.000
13.666	-.003	.001	.002	-.001	-.001
13.823	-.004	.001	.001	-.001	-.001
13.980	-.004	.001	.001	.000	-.001
14.137	-.004	.001	.000	.001	-.002
14.294	-.003	.001	-.000	.002	-.002
14.451	-.003	.001	-.001	.002	-.002
14.608	-.003	.001	-.001	.002	-.002
14.765	-.003	.001	-.001	.002	-.001
14.923	-.003	.001	-.001	.002	-.000
15.080	-.003	.001	-.001	.002	.001
15.237	-.003	.001	-.001	.001	.002
15.394	-.004	.001	-.001	.000	.003
15.551	-.004	.001	-.001	-.001	.005
15.708	-.004	.001	-.001	-.001	.005

TIME PRESSURE P(TIME) FOR DISCRETE VALUES OF Y

0.	.526E+06	.372E+06	-.760E-09	-.372E+06	-.526E+06	-.372E+06	-.822E-08	.372E+06
.125E-04	.427E+06	.570E+06	.379E+06	-.339E+05	-.427E+06	-.570E+06	-.379E+06	.339E+05
.250E-04	.221E+06	.603E+06	.631E+06	.290E+06	-.221E+06	-.603E+06	-.631E+06	-.290E+06
.375E-04	-.284E+06	.254E+06	.624E+06	.628E+06	.264E+06	-.254E+06	-.624E+06	-.628E+06
.500E-04	-.692E+06	-.321E+06	.238E+06	.658E+06	.692E+06	.321E+06	-.238E+06	-.658E+06
.625E-04	-.823E+06	-.843E+06	-.368E+06	.322E+06	.823E+06	.843E+06	.368E+06	-.322E+06
.750E-04	-.530E+06	-.102E+07	-.909E+06	-.268E+06	.530E+06	.102E+07	.909E+06	.268E+06
.875E-04	.298E+05	-.752E+06	-.109E+07	-.794E+06	-.298E+05	.752E+06	.109E+07	.794E+06
.100E-03	.576E+06	-.176E+06	-.825E+06	-.991E+06	-.576E+06	.176E+06	.825E+06	.991E+06
.113E-03	.802E+06	.397E+06	-.241E+06	-.737E+06	-.802E+06	-.397E+06	.241E+06	.737E+06
.125E-03	.609E+06	.679E+06	.351E+06	-.182E+06	-.609E+06	-.679E+06	-.351E+06	.182E+06
.138E-03	.109E+06	.537E+06	.651E+06	.383E+06	-.109E+06	-.537E+06	-.651E+06	-.383E+06
.150E-03	-.393E+06	.971E+05	.530E+06	.653E+06	.393E+06	-.971E+05	-.530E+06	-.653E+06
.163E-03	-.620E+06	.365E+06	.125E+06	.513E+06	.620E+06	.365E+06	-.125E+06	-.513E+06
.175E-03	-.466E+06	-.559E+06	-.344E+06	.721E+05	.466E+06	.559E+06	.344E+06	-.721E+05
.188E-03	.719E+04	-.373E+06	-.534E+06	-.383E+06	-.719E+04	.373E+06	.534E+06	.383E+06
.200E-03	.460E+06	.797E+05	-.347E+06	-.570E+06	-.460E+06	-.797E+05	.347E+06	.570E+06
.213E-03	.628E+06	.515E+06	.100E+06	-.373E+06	-.628E+06	-.515E+06	-.100E+06	.373E+06
.225E-03	.404E+06	.659E+06	.527E+06	.868E+05	-.404E+06	-.659E+06	-.527E+06	-.868E+05
.238E-03	-.847E+05	.408E+06	.661E+06	.527E+06	.847E+05	-.408E+06	-.661E+06	-.527E+06
.250E-03	-.551E+06	-.105E+06	.402E+06	.673E+06	.551E+06	.105E+06	-.402E+06	-.673E+06
.263E-03	-.706E+06	-.561E+06	-.115E+06	.418E+06	.706E+06	.561E+06	.115E+06	-.418E+06
.275E-03	-.457E+06	-.742E+06	-.593E+06	-.963E+05	.457E+06	.742E+06	.593E+06	.963E+05
.288E-03	.576E+05	.493E+06	-.755E+06	-.575E+06	-.576E+05	-.493E+06	.755E+06	.575E+06
.300E-03	.542E+06	.268E+05	-.504E+06	.740E+06	-.542E+06	-.268E+05	.504E+06	-.740E+06
.313E-03	.718E+06	.521E+06	.188E+05	-.494E+06	-.718E+06	-.521E+06	-.188E+05	.494E+06
.325E-03	.485E+06	.708E+06	.517E+06	.228E+05	-.485E+06	-.708E+06	-.517E+06	-.228E+05
.338E-03	-.205E+05	.486E+06	.708E+06	.515E+06	.205E+05	-.486E+06	-.708E+06	-.515E+06
.350E-03	-.504E+06	-.990E+04	.490E+06	.702E+06	.504E+06	.990E+04	-.490E+06	-.702E+06
.363E-03	-.685E+06	-.487E+06	-.449E+04	.481E+06	.685E+06	.487E+06	.449E+04	-.481E+06
.375E-03	-.462E+06	-.667E+06	-.481E+06	-.134E+05	.462E+06	.667E+06	.481E+06	.134E+05
.388E-03	.296E+05	-.447E+06	-.662E+06	-.489E+06	-.296E+05	.447E+06	.662E+06	.489E+06
.400E-03	.499E+06	.396E+05	-.443E+06	-.666E+06	-.499E+06	-.396E+05	.443E+06	.666E+06
.413E-03	.670E+06	.503E+06	.415E+05	-.444E+06	-.670E+06	-.503E+06	-.415E+05	.444E+06
.425E-03	.441E+06	.667E+06	.502E+06	.435E+05	-.441E+06	-.667E+06	-.502E+06	-.435E+05
.438E-03	-.525E+05	.433E+06	.664E+06	.507E+06	.525E+05	-.433E+06	-.664E+06	-.507E+06
.450E-03	-.519E+06	-.637E+05	.429E+06	.670E+06	.519E+06	.637E+05	-.429E+06	-.670E+06
.463E-03	-.663E+06	-.531E+06	-.678E+05	.435E+06	.663E+06	.531E+06	.678E+05	-.435E+06
.475E-03	-.445E+06	-.692E+06	-.534E+06	-.632E+05	.445E+06	.692E+06	.534E+06	.632E+05
.488E-03	.572E+05	-.451E+06	-.695E+06	-.532E+06	-.572E+05	.451E+06	.695E+06	.532E+06
.500E-03	.531E+06	.562E+05	-.451E+06	-.694E+06	-.531E+06	-.562E+05	.451E+06	.694E+06

TIME	UPSTREAM COMPONENT OF PRESSURE								
0.	.500E+06	.354E+06	-.858E-09	-.354E+06	-.500E+06	-.354E+06	-.809E-08	.354E+06	
.125E-04	.327E+06	.475E+06	.345E+06	.123E+05	-.327E+06	-.475E+06	-.345E+06	-.123E+05	
.250E-04	-.119E+05	.327E+06	.474E+06	.343E+06	.119E+05	-.327E+06	-.474E+06	-.343E+06	
.375E-04	-.349E+06	-.167E+05	.326E+06	.477E+06	.349E+06	.167E+05	-.326E+06	-.477E+06	
.500E-04	-.480E+06	-.352E+06	-.180E+05	.327E+06	.480E+06	.352E+06	.180E+05	-.327E+06	
.625E-04	-.333E+06	-.486E+06	-.354E+06	-.150E+05	.333E+06	.486E+06	.354E+06	.150E+05	
.750E-04	.112E+05	-.336E+06	-.487E+06	-.352E+06	-.112E+05	.336E+06	.487E+06	.352E+06	
.875E-04	.347E+06	.650E+04	-.338E+06	-.485E+06	-.347E+06	-.650E+04	.338E+06	.485E+06	
.100E-03	.482E+06	.344E+06	.547E+04	-.337E+06	-.482E+06	-.344E+06	-.547E+04	.337E+06	
.113E-03	.334E+06	.479E+06	.343E+06	.691E+04	-.334E+06	-.479E+06	-.343E+06	-.691E+04	
.125E-03	-.808E+04	.333E+06	.478E+06	.344E+06	.808E+04	-.333E+06	-.478E+06	-.344E+06	
.138E-03	-.345E+06	-.918E+04	.332E+06	.479E+06	.345E+06	.918E+04	-.332E+06	-.479E+06	
.150E-03	-.479E+06	-.345E+06	-.909E+04	.332E+06	.479E+06	.345E+06	.909E+04	-.332E+06	
.165E-03	.332E+06	-.478E+06	-.345E+06	-.922E+04	.332E+06	.478E+06	.345E+06	.922E+04	
.175E-03	.100E+05	-.331E+06	-.478E+06	-.345E+06	-.100E+05	.331E+06	.478E+06	.345E+06	
.188E-03	.346E+06	.107E+05	-.331E+06	-.478E+06	-.346E+06	-.107E+05	.331E+06	.478E+06	
.200E-03	.479E+06	.347E+06	.104E+05	-.331E+06	-.479E+06	-.347E+06	-.109E+05	.331E+06	
.215E-03	.332E+06	.480E+06	.347E+06	.108E+05	-.332E+06	-.480E+06	-.347E+06	-.108E+05	
.225E-03	-.107E+05	.332E+06	.480E+06	.347E+06	.107E+05	-.332E+06	-.480E+06	-.347E+06	
.238E-03	-.347E+06	-.110E+05	.332E+06	.480E+06	.347E+06	.110E+05	-.332E+06	-.480E+06	
.250E-03	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	.112E+05	-.332E+06	
.265E-03	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	.112E+05	
.275E-03	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	
.288E-03	.348E+06	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	
.300E-03	.480E+06	.348E+06	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	
.315E-03	.332E+06	.480E+06	.348E+06	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	
.325E-03	-.112E+05	.332E+06	.480E+06	.348E+06	.112E+05	-.332E+06	-.480E+06	-.348E+06	
.338E-03	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	.112E+05	-.332E+06	-.480E+06	
.350E-03	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	.112E+05	-.332E+06	
.365E-03	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	.112E+05	
.375E-03	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	
.388E-03	.348E+06	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	
.400E-03	.480E+06	.348E+06	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	
.415E-03	.332E+06	.480E+06	.348E+06	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	
.425E-03	-.112E+05	.332E+06	.480E+06	.348E+06	.112E+05	-.332E+06	-.480E+06	-.348E+06	
.438E-03	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	.112E+05	-.332E+06	-.480E+06	
.450E-03	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	.112E+05	-.332E+06	
.465E-03	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	.112E+05	
.475E-03	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	.348E+06	
.488E-03	.348E+06	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	.480E+06	
.500E-03	.480E+06	.348E+06	.112E+05	-.332E+06	-.480E+06	-.348E+06	-.112E+05	.332E+06	

PARAMETERS USED
 DTIME = .250E-05
 NY = 8
 NT = 201
 MAXCUN = 101
 CMAK = .800E+00
 RHOO = .100E+01
 C = .100E+04
 Y = .100E+00
 OMEGA0 = .500E+00

TEST R1STAR APPROACHES ITS ASYMPTOTIC VALUE

TIME STEP 1	R1STAR = 0.000	.143	ASYMPTOTE = 2.500	1.658
TIME STEP 6	R1STAR = -.166	.827	ASYMPTOTE = 1.675	2.489
TIME STEP 11	R1STAR = -.575	1.353	ASYMPTOTE = .595	2.944
TIME STEP 16	R1STAR = -1.143	1.608	ASYMPTOTE = -.575	2.944
TIME STEP 21	R1STAR = -1.750	1.526	ASYMPTOTE = -1.658	2.500
TIME STEP 26	R1STAR = -2.262	1.102	ASYMPTOTE = -2.489	1.675
TIME STEP 31	R1STAR = -2.554	.397	ASYMPTOTE = -2.944	.595
TIME STEP 36	R1STAR = -2.537	-.475	ASYMPTOTE = -2.944	-.575
TIME STEP 41	R1STAR = -2.171	-1.369	ASYMPTOTE = -2.500	-1.658
TIME STEP 46	R1STAR = -1.479	-2.124	ASYMPTOTE = -1.675	-2.489
TIME STEP 51	R1STAR = -.543	-2.602	ASYMPTOTE = -.595	-2.944
TIME STEP 56	R1STAR = .508	-2.703	ASYMPTOTE = .575	-2.944
TIME STEP 61	R1STAR = 1.518	-2.390	ASYMPTOTE = 1.658	-2.500
TIME STEP 66	R1STAR = 2.329	-1.695	ASYMPTOTE = 2.489	-1.675
TIME STEP 71	R1STAR = 2.807	-.714	ASYMPTOTE = 2.944	-.595
TIME STEP 76	R1STAR = 2.869	.406	ASYMPTOTE = 2.944	.575
TIME STEP 81	R1STAR = 2.493	1.492	ASYMPTOTE = 2.500	1.658
TIME STEP 86	R1STAR = 1.727	2.369	ASYMPTOTE = 1.675	2.489
TIME STEP 91	R1STAR = .682	2.894	ASYMPTOTE = .595	2.944
TIME STEP 96	R1STAR = -.484	2.977	ASYMPTOTE = -.575	2.944
TIME STEP 101	R1STAR = -1.592	2.597	ASYMPTOTE = -1.658	2.500
TIME STEP 106	R1STAR = -2.468	1.807	ASYMPTOTE = -2.489	1.675
TIME STEP 111	R1STAR = -2.972	.725	ASYMPTOTE = -2.944	.595
TIME STEP 116	R1STAR = -3.021	-.483	ASYMPTOTE = -2.944	-.575
TIME STEP 121	R1STAR = -2.603	-1.626	ASYMPTOTE = -2.500	-1.658
TIME STEP 126	R1STAR = -1.778	-2.527	ASYMPTOTE = -1.675	-2.489
TIME STEP 131	R1STAR = -.672	-3.040	ASYMPTOTE = -.595	-2.944
TIME STEP 136	R1STAR = .546	-3.083	ASYMPTOTE = .575	-2.944
TIME STEP 141	R1STAR = 1.686	-2.647	ASYMPTOTE = 1.658	-2.500
TIME STEP 146	R1STAR = 2.570	-1.797	ASYMPTOTE = 2.489	-1.675
TIME STEP 151	R1STAR = 3.061	-.664	ASYMPTOTE = 2.944	-.595
TIME STEP 156	R1STAR = 3.079	.576	ASYMPTOTE = 2.944	.575
TIME STEP 161	R1STAR = 2.621	1.730	ASYMPTOTE = 2.500	1.658
TIME STEP 166	R1STAR = 1.757	2.617	ASYMPTOTE = 1.675	2.489
TIME STEP 171	R1STAR = .620	3.101	ASYMPTOTE = .595	2.944
TIME STEP 176	R1STAR = -.615	3.105	ASYMPTOTE = -.575	2.944
TIME STEP 181	R1STAR = -1.756	2.628	ASYMPTOTE = -1.658	2.500
TIME STEP 186	R1STAR = -2.626	1.746	ASYMPTOTE = -2.489	1.675
TIME STEP 191	R1STAR = -3.092	.593	ASYMPTOTE = -2.944	.595
TIME STEP 196	R1STAR = -3.082	-.649	ASYMPTOTE = -2.944	-.575
TIME STEP 201	R1STAR = -2.597	-1.790	ASYMPTOTE = -2.500	-1.658

TIME	INPUT	R(TIME)
0.	0.000	.101 .143 .101 -.000 -.101 -.143 -.101
.125E-04	-.166 .468 .827 .702 .166 -.468 -.827 -.702	
.250E-04	-.575 .550 1.353 1.363 .575 -.550 -1.353 -1.363	
.375E-04	-1.143 .329 1.608 1.945 1.143 -.329 -1.608 -1.945	
.500E-04	-1.750 -.159 1.526 2.316 1.750 .159 -1.526 -2.316	
.625E-04	-2.262 -.820 1.102 2.379 2.262 .820 -1.102 -2.379	
.750E-04	-2.554 -1.525 .397 2.087 2.554 1.525 -.397 -2.087	
.875E-04	-2.537 -2.130 -.475 1.458 2.537 2.130 .475 -1.458	
1.00E-03	-2.171 -2.503 -1.369 .567 2.171 2.503 1.369 -.567	
.113E-03	-1.479 -2.548 -2.124 -.456 1.479 2.548 2.124 .456	
.125E-03	-.543 -2.224 -2.602 -1.456 .543 2.224 2.602 1.456	
.138E-03	.508 -1.552 -2.703 -2.271 -.508 1.552 2.703 2.271	
.150E-03	1.518 -.617 -2.390 -2.764 -1.518 .617 2.390 2.764	
.163E-03	2.329 .448 -1.695 -2.845 -2.329 -.448 1.695 2.845	
.175E-03	2.807 1.480 -.714 -2.490 -2.807 -1.480 .714 2.490	
.188E-03	2.869 2.316 .406 -1.741 -2.869 -2.316 -.406 1.741	
.200E-03	2.493 2.818 1.492 -.708 -2.493 -2.818 -1.492 .708	
.213E-03	1.727 2.896 2.369 .454 -1.727 -2.896 -2.369 -.454	
.225E-03	.682 2.529 2.894 1.564 -.682 -2.529 -2.894 -1.564	
.238E-03	-.484 1.763 2.977 2.448 .484 -1.763 -2.977 -2.448	
.250E-03	-1.592 .711 2.597 2.962 1.592 -.711 -2.597 -2.962	
.263E-03	-2.468 -.468 1.807 3.023 2.468 .468 -1.807 -3.023	
.275E-03	-2.972 -1.589 .725 2.614 2.972 1.589 -.725 -2.614	
.288E-03	-3.021 -2.478 -.483 1.795 3.021 2.478 .483 -1.795	
.300E-03	-2.603 -2.991 -1.626 .690 2.603 2.991 1.626 -.690	
.313E-03	-1.778 -3.044 -2.527 -.529 1.778 3.044 2.527 .529	
.325E-03	-.672 -2.624 -3.040 -1.675 .672 2.624 3.040 1.675	
.338E-03	.546 -1.794 -3.083 -2.566 -.546 1.794 3.083 2.566	
.350E-03	1.686 -.680 -2.647 -3.064 -1.686 .680 2.647 3.064	
.363E-03	2.570 .547 -1.797 -3.088 -2.570 -.547 1.797 3.088	
.375E-03	3.061 1.695 -.664 -2.634 -3.061 -1.695 .664 2.634	
.388E-03	3.079 2.584 .576 -1.770 -3.079 -2.584 -.576 1.770	
.400E-03	2.621 3.076 1.730 -.630 -2.621 -3.076 -1.730 .630	
.413E-03	1.757 3.093 2.617 .608 -1.757 -3.093 -2.617 -.608	
.425E-03	.620 2.631 3.101 1.755 -.620 -2.631 -3.101 -1.755	
.438E-03	-.615 1.761 3.105 2.630 .615 -1.761 -3.105 -2.630	
.450E-03	-1.756 .617 2.628 3.100 1.756 -.617 -2.628 -3.100	
.463E-03	-2.526 -.623 1.746 3.091 2.526 .623 -1.746 -3.091	
.475E-03	-3.092 -1.767 .593 2.606 3.092 1.767 -.593 -2.606	
.488E-03	-3.082 -2.638 -.649 1.720 3.082 2.638 .649 -1.720	
.500E-03	-2.597 -3.102 -1.790 .570 2.597 3.102 1.790 -.570	

TIME	INPUT	Q(TIME)
0.	1.000	.707 -.000 -.707 -1.000 -.707 -.000 .707
.125E-04	.924 .924 .383 -.383 -.924 -.924 -.383 .383	
.250E-04	.707 1.000 .707 .000 -.707 -1.000 -.707 -.000	
.375E-04	.383 .924 .924 .383 -.383 -.924 -.924 -.383	
.500E-04	.000 .707 1.000 .707 -.000 -.707 -1.000 -.707	
.625E-04	-.383 .383 .924 .924 .383 -.383 -.924 -.924	
.750E-04	-.707 .000 .707 1.000 .707 -.000 -.707 -1.000	
.875E-04	-.924 -.383 .383 .924 .924 .383 -.383 -.924	
1.00E-03	-1.000 -.707 .000 .707 1.000 .707 .000 -.707	
.113E-03	-.924 -.924 -.383 .383 .924 .924 .383 -.383	
.125E-03	-.707 -1.000 -.707 .000 .707 1.000 .707 .000	
.138E-03	-.383 -.924 -.924 -.383 .383 .924 .924 .383	
.150E-03	-.000 -.707 -1.000 -.707 .000 .707 1.000 .707	
.163E-03	.383 -.383 -.924 -.924 -.383 .383 .924 .924	
.175E-03	.707 -.000 -.707 -1.000 -.707 .000 .707 1.000	
.188E-03	.924 .383 -.383 -.924 -.924 -.383 .383 .924	
.200E-03	1.000 .707 -.000 -.707 -1.000 -.707 .000 .707	
.213E-03	.924 .924 .383 -.383 -.924 -.924 -.383 .383	
.225E-03	.707 1.000 .707 -.000 -.707 -1.000 -.707 .000	
.238E-03	.383 .924 .924 .383 -.383 -.924 -.924 -.383	
.250E-03	.000 .707 1.000 .707 -.000 -.707 -1.000 -.707	
.263E-03	-.383 .383 .924 .924 .383 -.383 -.924 -.924	
.275E-03	-.707 .000 .707 1.000 .707 -.000 -.707 -1.000	
.288E-03	-.924 -.383 .383 .924 .924 .383 -.383 -.924	
.300E-03	-1.000 -.707 .000 .707 1.000 .707 .000 -.707	
.313E-03	-.924 -.924 -.383 .383 .924 .924 .383 -.383	
.325E-03	-.707 -1.000 -.707 .000 .707 1.000 .707 .000	
.338E-03	-.383 -.924 -.924 -.383 .383 .924 .924 .383	
.350E-03	-.000 -.707 -1.000 -.707 .000 .707 1.000 .707	
.363E-03	.383 -.383 .924 -.924 -.383 .383 .924 .924	
.375E-03	.707 -.000 -.707 -1.000 -.707 .000 .707 1.000	
.388E-03	.924 .383 -.383 -.924 -.924 -.383 .383 .924	
.400E-03	1.000 .707 -.000 -.707 -1.000 -.707 .000 .707	
.413E-03	.924 .924 .383 -.383 -.924 -.924 -.383 .383	
.425E-03	.707 1.000 .707 -.000 -.707 -1.000 -.707 .000	
.438E-03	.383 .924 .924 .383 -.383 -.924 -.924 -.383	
.450E-03	.000 .707 1.000 .707 -.000 -.707 -1.000 -.707	
.463E-03	-.383 .383 .924 .924 .383 -.383 -.924 -.924	
.475E-03	-.707 .000 .707 1.000 .707 -.000 -.707 -1.000	
.488E-03	-.924 -.383 .383 .924 .924 .383 -.383 -.924	
.500E-03	-1.000 -.707 .000 .707 1.000 .707 -.000 -.707	

TIME	GREENS	FUNCTION	HR		
0,000	25,465	-.000	-.000	-.000	0,000
.157	-1,581	.078	.284	.317	.324
.314	-2,305	.183	.370	.398	.404
.471	-2,630	.332	.389	.395	.397
.628	-2,648	.508	.341	.318	.314
.785	-2,408	.578	.240	.194	.185
.942	-1,999	.804	.111	.060	.050
1,100	-1,525	.855	-.016	-.049	-.055
1,257	-1,081	.815	-.113	-.108	-.107
1,414	-.731	.688	-.159	-.111	-.104
1,571	-.505	.497	-.146	-.069	-.058
1,728	-.395	.280	-.077	-.007	.003
1,885	-.367	.076	.030	.051	.054
2,042	-.379	-.083	.153	.082	.076
2,199	-.392	-.180	.267	.077	.064
2,356	-.384	-.213	.351	.040	.028
2,513	-.348	-.197	.390	-.011	-.016
2,670	-.293	-.155	.382	-.056	-.047
2,827	-.236	-.110	.332	-.078	-.054
2,985	-.191	-.078	.255	-.064	-.034
3,142	-.167	-.067	.167	-.018	.003
3,299	-.162	-.075	.085	.052	.039
3,456	-.170	-.091	.020	.126	.059
3,613	-.179	-.106	-.020	.189	.053
3,770	-.181	-.111	-.037	.228	.021
3,927	-.172	-.103	-.036	.237	-.023
4,084	-.153	-.087	-.025	.219	-.061
4,241	-.132	-.067	-.013	.183	-.075
4,398	-.114	-.051	-.006	.139	-.051
4,555	-.104	-.044	-.006	.097	.011
4,712	-.104	-.045	-.014	.061	.100
4,869	-.110	-.052	-.026	.034	.197
5,027	-.117	-.060	-.037	.015	.280
5,184	-.120	-.064	-.043	.002	.329
5,341	-.116	-.060	-.043	-.007	.336
5,498	-.106	-.051	-.036	-.011	.303
5,655	-.093	-.040	-.027	-.008	.242
5,812	-.083	-.030	-.017	.004	.170
5,969	-.078	-.025	-.012	.024	.104
6,126	-.079	-.026	-.012	.050	.056
6,283	-.085	-.032	-.018	.077	.031
6,440	-.091	-.038	-.028	.099	.025
6,597	-.094	-.042	-.037	.111	.030
6,754	-.092	-.041	-.042	.110	.037
6,912	-.084	-.034	-.039	.095	.039
7,069	-.075	-.025	-.027	.072	.034
7,226	-.067	-.016	-.008	.046	.022
7,383	-.064	-.011	.017	.022	.008
7,540	-.066	-.013	.041	.006	.004
7,697	-.072	-.020	.061	-.001	-.010
7,854	-.079	-.029	.073	-.000	-.009

TIME	GREENS	FUNCTION	HR		
8,011	-.083	-.036	.075	.004	-.005
8,168	-.081	-.037	.069	.009	.000
8,325	-.074	-.030	.055	.010	.003
8,482	-.064	-.014	.038	.008	.002
8,639	-.055	.006	.021	.002	-.003
8,796	-.052	.027	.009	-.005	-.009
8,954	-.057	.045	.001	-.011	-.014
9,111	-.067	.056	-.001	-.013	-.016
9,268	-.079	.059	.000	-.012	-.015
9,425	-.087	.055	.003	-.008	-.011
9,582	-.085	.046	.005	-.005	-.008
9,739	-.071	.035	.006	-.003	-.005
9,896	-.044	.024	.004	-.003	-.005
10,053	-.009	.015	-.000	-.006	-.008
10,210	.025	.009	-.005	-.010	-.012
10,367	.053	.003	-.008	-.014	-.015
10,524	.067	-.000	-.010	-.015	-.016
10,681	.066	-.003	-.010	-.014	-.015
10,838	.052	-.003	-.007	-.010	-.012
10,996	.029	.000	-.004	-.007	-.008
11,153	.005	.007	-.002	-.005	-.006
11,310	-.015	.017	-.001	-.005	-.006
11,467	-.011	.041	-.003	-.020	-.026
11,624	-.021	.048	-.007	-.019	-.024
11,781	-.020	.052	-.011	-.018	-.020
11,938	-.026	.053	-.015	-.016	-.017
12,095	-.024	.050	-.015	-.015	-.016
12,252	-.021	.045	-.012	-.014	-.016
12,409	-.020	.038	-.005	-.014	-.018
12,566	-.021	.031	.005	-.015	-.021
12,723	-.024	.024	.016	-.017	-.024
12,881	-.029	.019	.027	-.019	-.025
13,038	-.034	.015	.034	-.020	-.025
13,195	-.037	.013	.038	-.021	-.023
13,352	-.023	.013	.021	-.020	-.004
13,509	-.027	.013	.020	-.018	-.003
13,666	-.030	.013	.017	-.013	-.005
13,823	-.032	.013	.013	-.007	-.008
13,980	-.033	.013	.008	.001	-.012
14,137	-.032	.012	.003	.008	-.016
14,294	-.031	.011	-.001	.015	-.019
14,451	-.029	.009	-.005	.020	-.020
14,608	-.028	.008	-.008	.022	-.018
14,765	-.027	.007	-.009	.022	-.012
14,923	-.027	.006	-.010	.019	-.003
15,080	-.028	.006	-.009	.014	.007
15,237	-.030	.006	-.008	.007	.020
15,394	-.032	.007	-.007	.001	.031
15,551	-.034	.007	-.006	-.005	.042
15,708	-.035	.007	-.006	-.009	.049

TIME	GREENS	FUNCTION	JR
0.000	-3.698	-1.920	-.629
.157	-3.114	-1.248	-.514
.314	-2.208	-.851	-.348
.471	-1.138	-.420	-.171
.628	-.078	-.054	-.024
.785	.820	.185	.072
.942	1.448	.264	.098
1.100	1.764	.198	.073
1.257	1.788	.050	.020
1.414	1.590	-.108	-.032
1.571	1.267	-.208	-.059
1.728	.912	-.203	-.055
1.885	.601	-.090	-.024
2.042	.375	.105	.017
2.199	.242	.329	.046
2.356	.184	.525	.050
2.513	.172	.650	.024
2.670	.174	.681	-.020
2.827	.170	.625	-.062
2.985	.152	.510	-.080
3.142	.123	.373	-.056
3.299	.091	.249	.012
3.456	.065	.163	.114
3.613	.051	.121	.228
3.770	.048	.113	.325
3.927	.052	.122	.383
4.084	.055	.131	.390
4.241	.054	.128	.348
4.398	.048	.111	.273
4.555	.037	.085	.188
4.712	.027	.059	.117
4.869	.020	.043	.075
5.027	.019	.039	.066
5.184	.021	.046	.082
5.341	.025	.055	.108
5.498	.027	.061	.125
5.655	.026	.058	.120
5.812	.021	.045	.089
5.969	.014	.029	.036
6.126	.009	.016	-.027
6.283	.007	.011	-.086
6.440	.009	.017	-.127
6.597	.013	.031	-.145
6.754	.018	.045	-.139
6.912	.019	.051	-.116
7.069	.015	.042	-.086
7.226	.009	.016	-.057
7.383	.003	-.023	-.037
7.540	-.001	-.066	-.028
7.697	.001	-.103	-.027
7.854	.008	-.125	-.030

TIME	GREENS	FUNCTION	JR
8.011	.018	-.129	-.033
8.168	.024	-.116	-.033
8.325	.023	-.093	-.029
8.482	.011	-.066	-.023
8.639	-.013	-.044	-.017
8.796	-.045	-.032	-.013
8.954	-.077	-.029	-.012
9.111	-.101	-.032	-.014
9.268	-.113	-.037	-.016
9.425	-.109	-.039	-.016
9.582	-.093	-.037	-.016
9.739	-.069	-.030	-.013
9.896	-.046	-.022	-.010
10.053	-.029	-.016	-.008
10.210	-.023	-.014	-.007
10.367	-.027	-.016	-.007
10.524	-.037	-.020	-.009
10.681	-.046	-.023	-.010
10.838	-.049	-.024	-.011
10.996	-.041	-.020	-.009
11.153	-.024	-.015	-.007
11.310	.000	-.009	-.005
11.467	.019	-.015	-.010
11.624	.037	-.020	-.013
11.781	.050	-.023	-.013
11.938	.057	-.023	-.011
12.095	.056	-.020	-.006
12.252	.048	-.012	-.001
12.409	.037	-.002	.002
12.566	.025	.011	.003
12.723	.015	.024	.001
12.881	.008	.036	-.004
13.038	.006	.045	-.010
13.195	.007	.050	-.015
13.352	.004	.051	-.010
13.509	.004	.048	-.003
13.666	.005	.042	.006
13.823	.008	.034	.016
13.980	.011	.026	.025
14.137	.013	.019	.033
14.294	.014	.014	.039
14.451	.013	.011	.042
14.608	.011	.011	.042
14.765	.008	.012	.041
14.923	.005	.014	.039
15.080	.003	.016	.035
15.237	.001	.017	.032
15.394	.001	.016	.028
15.551	.001	.015	.025
15.708	.003	.012	.022

TIME	GREENS	FUNCTION	KR			TIME	GREENS	FUNCTION	KR		
0.000	25.465	.000	.000	-.000	0.000	8.011	-.009	-.004	.008	.000	-.001
.157	-.187	.009	.032	.035	.036	8.168	-.009	-.004	.008	.001	.000
.314	-.256	.020	.041	.044	.045	8.325	-.008	-.003	.006	.001	.000
.471	-.292	.037	.043	.044	.044	8.482	-.007	-.002	.004	.001	.000
.628	-.294	.056	.038	.035	.035	8.639	-.006	.001	.002	.000	.000
.785	-.268	.075	.027	.022	.021	8.796	-.006	.003	.001	-.001	.001
.942	-.222	.089	.012	.007	.006	8.954	-.006	.005	.000	-.001	.002
1.100	-.169	.095	-.002	-.005	-.006	9.111	-.007	.006	-.000	-.001	.002
1.257	-.120	.091	-.013	-.012	-.012	9.268	-.009	.007	.000	-.001	.002
1.414	-.081	.076	-.018	-.012	-.012	9.425	-.010	.006	.000	-.001	.002
1.571	-.056	.055	-.016	-.008	-.006	9.582	-.009	.005	.001	-.001	.001
1.728	-.044	.031	-.009	-.001	.000	9.739	-.008	.004	.001	-.000	.001
1.885	-.041	.008	.003	.006	.006	9.896	-.005	.003	.000	-.000	.001
2.042	-.042	-.009	.017	.009	.008	10.053	-.001	.002	-.000	-.001	.001
2.199	-.044	-.020	.030	.009	.007	10.210	.003	.001	-.001	-.001	.001
2.356	-.043	-.024	.039	.004	.003	10.367	.006	.000	-.001	-.002	.002
2.513	-.039	-.022	.043	-.001	-.002	10.524	.007	-.000	-.001	-.002	.002
2.670	-.033	-.017	.042	-.006	-.005	10.681	.007	-.000	-.001	-.002	.002
2.827	-.026	-.012	.037	-.009	-.006	10.838	.006	-.000	-.001	-.001	.001
2.985	-.021	-.009	.028	-.007	-.004	10.996	.003	.000	-.000	-.001	.001
3.142	-.019	-.007	.019	-.002	.000	11.153	.001	.001	-.000	-.001	.001
3.299	-.018	-.008	.009	.006	.004	11.310	-.002	.002	-.000	-.001	.001
3.456	-.019	-.010	.002	.014	.007	11.467	-.001	.005	-.000	-.002	.003
3.613	-.020	-.012	-.002	.021	.006	11.624	-.002	.005	-.001	-.002	.003
3.770	-.020	-.012	-.004	.025	.002	11.781	-.003	.006	-.001	-.002	.002
3.927	-.019	-.011	-.004	.026	-.003	11.938	-.003	.006	-.002	-.002	.002
4.084	-.017	-.010	-.003	.024	-.007	12.095	-.003	.006	-.002	-.002	.002
4.241	-.015	-.007	-.001	.020	-.008	12.252	-.002	.005	-.001	-.002	.002
4.398	-.013	-.006	-.001	.015	-.006	12.409	-.002	.004	-.001	-.002	.002
4.555	-.012	-.005	-.001	.011	.001	12.566	-.002	.003	.001	-.002	.002
4.712	-.012	-.005	-.002	.007	.011	12.723	-.003	.003	.002	-.002	.003
4.869	-.012	-.006	-.003	.004	.022	12.881	-.003	.002	.003	-.002	.003
5.027	-.013	-.007	-.004	.002	.031	13.038	-.004	.002	.004	-.002	.003
5.184	-.013	-.007	-.005	.000	.037	13.195	-.004	.001	.004	-.002	.003
5.341	-.013	-.007	-.005	-.001	.037	13.352	-.003	.001	.002	-.002	.000
5.498	-.012	-.006	-.004	-.001	.034	13.509	-.003	.001	.002	-.002	.000
5.655	-.010	-.004	-.003	-.001	.027	13.666	-.003	.001	.002	-.001	.001
5.812	-.009	-.003	-.002	.000	.019	13.823	-.004	.001	.001	-.001	.001
5.969	-.009	-.003	-.001	.003	.012	13.980	-.004	.001	.001	.000	.001
6.126	-.009	-.003	-.001	.006	.006	14.137	-.004	.001	.000	.001	.002
6.283	-.009	-.004	-.002	.009	.003	14.294	-.003	.001	-.000	.002	.002
6.440	-.010	-.004	-.003	.011	.003	14.451	-.003	.001	-.001	.002	.002
6.597	-.010	-.005	-.004	.012	.003	14.608	-.003	.001	-.001	.002	.002
6.754	-.010	-.005	-.005	.012	.004	14.765	-.003	.001	-.001	.002	.001
6.912	-.009	-.004	-.004	.011	.004	14.923	-.003	.001	-.001	.002	.000
7.069	-.008	-.003	-.003	.008	.004	15.080	-.003	.001	-.001	.002	.001
7.226	-.007	-.002	-.001	.005	.002	15.237	-.003	.001	-.001	.001	.002
7.383	-.007	-.001	.002	.002	.001	15.394	-.004	.001	-.001	.000	.003
7.540	-.007	-.001	.005	.001	-.000	15.551	-.004	.001	-.001	-.001	.005
7.697	-.008	-.002	.007	-.000	-.001	15.708	-.004	.001	-.001	-.001	.005
7.854	-.009	-.003	.008	-.000	-.001						

TIME	PRESSURE P(TIME) FOR DISCRETE VALUES OF Y							
0.	.526E+06	.372E+06	-.760E-09	-.372E+06	-.526E+06	-.372E+06	-.822E+08	.372E+06
.125E-04	.542E+06	.528E+06	.205E+06	-.238E+06	-.542E+06	-.528E+06	-.205E+06	.238E+06
.250E-04	.620E+06	.741E+06	.428E+06	-.135E+06	-.620E+06	-.741E+06	-.428E+06	.135E+06
.375E-04	.440E+06	.766E+06	.644E+06	.144E+06	-.440E+06	-.766E+06	-.644E+06	-.144E+06
.500E-04	.926E+05	.599E+06	.755E+06	.468E+06	-.926E+05	-.599E+06	-.755E+06	-.468E+06
.625E-04	-.377E+06	.235E+06	.709E+06	.767E+06	.377E+06	-.235E+06	-.709E+06	-.767E+06
.750E-04	-.839E+06	-.256E+06	.476E+06	.930E+06	.839E+06	.256E+06	-.476E+06	-.930E+06
.875E-04	-.121E+07	-.798E+06	.798E+05	.911E+06	.121E+07	.798E+06	-.798E+05	-.911E+06
.100E-03	-.158E+07	-.126E+07	-.427E+06	.673E+06	.138E+07	.128E+07	.427E+06	-.673E+06
.115E-03	-.131E+07	-.160E+07	-.955E+06	.252E+06	.131E+07	.160E+07	.955E+06	-.252E+06
.125E-03	-.963E+06	-.169E+07	-.141E+07	-.301E+06	.983E+06	.169E+07	.141E+07	.301E+06
.138E-03	-.447E+06	-.151E+07	-.169E+07	-.80E+06	.447E+06	.151E+07	.169E+07	.80E+06
.150E-03	.222E+06	-.107E+07	-.174E+07	-.139E+07	-.222E+06	.107E+07	.174E+07	.139E+07
.165E-03	.904E+06	-.433E+06	-.152E+07	-.71E+07	-.904E+06	.433E+06	.152E+07	.171E+07
.175E-03	.149E+07	.312E+06	-.104E+07	-.179E+07	-.149E+07	-.312E+06	.104E+07	.179E+07
.188E-03	.186E+07	.104E+07	-.384E+06	-.158E+07	-.186E+07	-.104E+07	.384E+06	.158E+07
.200E-03	.194E+07	.163E+07	.369E+06	-.111E+07	-.194E+07	-.163E+07	-.369E+06	.111E+07
.214E-03	.172E+07	.199E+07	.109E+07	-.439E+06	-.172E+07	-.199E+07	-.109E+07	.439E+06
.225E-03	.120E+07	.203E+07	.167E+07	.336E+06	-.120E+07	-.203E+07	-.167E+07	-.336E+06
.238E-03	.470E+06	.175E+07	.201E+07	.109E+07	-.470E+06	-.175E+07	-.201E+07	-.109E+07
.250E-03	-.358E+06	.118E+07	.203E+07	.169E+07	.358E+06	-.118E+07	-.203E+07	-.169E+07
.265E-03	-.114E+07	.419E+06	.173E+07	.203E+07	.114E+07	-.419E+06	-.173E+07	-.203E+07
.275E-03	-.176E+07	-.425E+06	.116E+07	.206E+07	.176E+07	.425E+06	-.116E+07	-.206E+07
.288E-03	-.211E+07	-.122E+07	.391E+06	.177E+07	.211E+07	.122E+07	-.391E+06	-.177E+07
.300E-03	-.213E+07	-.183E+07	-.453E+06	.119E+07	.213E+07	.183E+07	.453E+06	.119E+07
.315E-03	-.192E+07	-.216E+07	-.124E+07	.412E+06	.182E+07	.216E+07	.124E+07	-.412E+06
.325E-03	-.122E+07	-.217E+07	-.164E+07	-.441E+06	.122E+07	.217E+07	.164E+07	.441E+06
.338E-03	-.419E+06	-.183E+07	-.217E+07	.124E+07	.419E+06	.183E+07	.217E+07	.124E+07
.350E-03	.456E+06	-.121E+07	-.216E+07	-.185E+07	-.456E+06	.121E+07	.216E+07	.185E+07
.365E-03	.127E+07	-.392E+06	-.182E+07	-.218E+07	-.127E+07	.392E+06	.182E+07	.218E+07
.375E-03	.189E+07	.490E+06	-.119E+07	-.218E+07	-.189E+07	-.490E+06	.119E+07	.218E+07
.388E-03	.222E+07	.130E+07	-.378E+06	-.184E+07	-.222E+07	-.130E+07	.378E+06	.184E+07
.400E-03	.221E+07	.191E+07	.501E+06	-.121E+07	-.221E+07	-.191E+07	-.501E+06	.121E+07
.415E-03	.185E+07	.223E+07	.131E+07	-.353E+06	-.185E+07	-.223E+07	-.131E+07	.353E+06
.425E-03	.120E+07	.221E+07	.192E+07	.503E+06	-.120E+07	-.221E+07	-.192E+07	-.503E+06
.435E-03	.370E+06	.164E+07	.223E+07	.131E+07	-.370E+06	-.164E+07	-.223E+07	-.131E+07
.450E-03	-.524E+06	.119E+07	.220E+07	.193E+07	.524E+06	-.119E+07	-.220E+07	-.193E+07
.465E-03	-.134E+07	.347E+06	.183E+07	.224E+07	.134E+07	-.347E+06	-.183E+07	-.224E+07
.475E-03	-.195E+07	-.346E+06	.118E+07	.221E+07	.195E+07	.346E+06	-.118E+07	-.221E+07
.488E-03	-.226E+07	-.136E+07	.340E+06	.184E+07	-.226E+07	-.136E+07	-.340E+06	-.184E+07
.500E-03	-.222E+07	-.196E+07	-.550E+06	.118E+07	.222E+07	.196E+07	.550E+06	-.118E+07

TIME	UPSTREAM COMPONENT OF PRESSURE							
0.	.500E+06	.354E+06	-.858E-09	-.354E+06	-.500E+06	-.354E+06	-.809E-08	.354E+06
.125E-04	.434E+06	.439E+06	.187E+06	-.175E+06	-.434E+06	-.439E+06	-.187E+06	.175E+06
.250E-04	.330E+06	.472E+06	.338E+06	.595E+04	-.330E+06	-.472E+06	-.338E+06	-.595E+04
.375E-04	.169E+06	.429E+06	.438E+06	.190E+06	-.169E+06	-.429E+06	-.438E+06	-.190E+06
.500E-04	-.145E+05	.321E+06	.469E+06	.342E+06	.145E+05	-.321E+06	-.469E+06	-.342E+06
.625E-04	-.199E+06	.161E+06	.427E+06	.442E+06	.199E+06	-.161E+06	-.427E+06	-.442E+06
.750E-04	-.351E+06	-.240E+05	.317E+06	.473E+06	.351E+06	.240E+05	-.317E+06	-.473E+06
.875E-04	-.451E+06	-.207E+06	.158E+06	.431E+06	.451E+06	.207E+06	-.158E+06	-.431E+06
.100E-03	-.480E+06	-.359E+06	-.273E+05	.320E+06	.480E+06	.359E+06	.273E+05	-.320E+06
.113E-03	-.437E+06	-.457E+06	-.210E+06	.161E+06	.437E+06	.457E+06	.210E+06	-.161E+06
.125E-03	-.324E+06	-.485E+06	-.361E+06	.258E+05	.324E+06	.485E+06	.361E+06	-.258E+05
.138E-03	-.163E+06	-.439E+06	-.458E+06	-.209E+06	.163E+06	.439E+06	.458E+06	.209E+06
.150E-03	.251E+05	-.325E+06	-.485E+06	-.361E+06	-.251E+05	.325E+06	.485E+06	.361E+06
.163E-03	.209E+06	-.162E+06	-.439E+06	-.458E+06	-.209E+06	.162E+06	.439E+06	.458E+06
.175E-03	.362E+06	.260E+05	-.325E+06	-.486E+06	-.362E+06	-.260E+05	.325E+06	.486E+06
.188E-03	.459E+06	.210E+06	-.162E+06	-.439E+06	-.459E+06	-.210E+06	.162E+06	.439E+06
.200E-03	.487E+06	.363E+06	.265E+05	-.326E+06	-.487E+06	-.363E+06	-.265E+05	.326E+06
.213E-03	.440E+06	.461E+06	.211E+06	-.162E+06	-.440E+06	-.461E+06	-.211E+06	.162E+06
.225E-03	.326E+06	.488E+06	.364E+06	.264E+05	-.326E+06	-.488E+06	-.364E+06	-.264E+05
.238E-03	.162E+06	.440E+06	.461E+06	.211E+06	-.162E+06	-.440E+06	-.461E+06	-.211E+06
.250E-03	-.272E+05	.325E+06	.487E+06	.364E+06	.272E+05	-.325E+06	-.487E+06	-.364E+06
.263E-03	-.212E+06	.161E+06	.440E+06	.461E+06	.212E+06	-.161E+06	-.440E+06	-.461E+06
.275E-03	-.364E+06	-.272E+05	.325E+06	.487E+06	.364E+06	.272E+05	-.325E+06	-.487E+06
.288E-03	-.461E+06	-.212E+06	.161E+06	.440E+06	.461E+06	.212E+06	-.161E+06	-.440E+06
.300E-03	-.487E+06	-.364E+06	-.272E+05	.325E+06	-.487E+06	-.364E+06	-.272E+05	.325E+06
.313E-03	-.440E+06	-.461E+06	-.212E+06	.161E+06	.440E+06	.461E+06	.212E+06	-.161E+06
.325E-03	-.325E+06	-.487E+06	-.364E+06	-.272E+05	.325E+06	.487E+06	.364E+06	.272E+05
.338E-03	-.161E+06	-.440E+06	.461E+06	-.212E+06	.161E+06	.440E+06	.461E+06	.212E+06
.350E-03	.272E+05	-.325E+06	-.487E+06	-.364E+06	-.272E+05	.325E+06	.487E+06	.364E+06
.363E-03	.212E+06	-.161E+06	-.440E+06	-.461E+06	-.212E+06	.161E+06	.440E+06	.461E+06
.375E-03	.364E+06	.272E+05	-.325E+06	-.487E+06	-.364E+06	-.272E+05	.325E+06	.487E+06
.388E-03	.461E+06	.212E+06	-.161E+06	-.440E+06	.461E+06	.212E+06	-.161E+06	-.440E+06
.400E-03	.487E+06	.364E+06	.272E+05	-.325E+06	-.487E+06	-.364E+06	-.272E+05	.325E+06
.413E-03	.440E+06	.461E+06	.212E+06	-.161E+06	-.440E+06	-.461E+06	-.212E+06	.161E+06
.425E-03	.325E+06	.487E+06	.364E+06	.272E+05	-.325E+06	-.487E+06	-.364E+06	-.272E+05
.438E-03	.161E+06	.440E+06	.461E+06	.212E+06	-.161E+06	-.440E+06	-.461E+06	-.212E+06
.450E-03	-.272E+05	.325E+06	.487E+06	.364E+06	.272E+05	-.325E+06	-.487E+06	-.364E+06
.463E-03	-.212E+06	.161E+06	.440E+06	.461E+06	.212E+06	-.161E+06	-.440E+06	-.461E+06
.475E-03	-.364E+06	-.272E+05	.325E+06	.487E+06	.364E+06	.272E+05	-.325E+06	-.487E+06
.488E-03	-.461E+06	-.212E+06	.161E+06	.440E+06	.461E+06	.212E+06	-.161E+06	-.440E+06
.500E-03	-.487E+06	-.364E+06	-.272E+05	.325E+06	-.487E+06	-.364E+06	-.272E+05	.325E+06

REFERENCES

1. Erdos, John I., and Alzner, Edgar, "Computation of Unsteady Transonic Flows Through Rotating and Stationary Cascades, I - Method of Analysis," NASA CR-2900, 1977.
2. Alzner, Edgar, and Kalben, Paul P., "Computation of Unsteady Transonic Flows Through Rotating and Stationary Cascades, II - User's Guide to FORTRAN Program B2DATL," NASA CR-2901, 1977.

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16. Abstract <p>A small perturbation type analysis has been developed for the acoustic far field in an infinite duct extending upstream and downstream of an axial turbomachinery stage. The analysis is designed to interface with the numerical solution of the near field of the blade rows described in NASA CR-2900 and thereby to provide the necessary closure condition to complete the statement of infinite duct boundary conditions for the subject problem. The present analysis differs from conventional inlet duct analyses in that a simple harmonic time dependence was not assumed, since a transient signal is generated by the numerical near-field solution and periodicity is attained only asymptotically. A description of the computer code developed to carry out the necessary convolutions numerically is included, as well as the results of a sample application using an impulsively initiated harmonic signal.</p>			
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