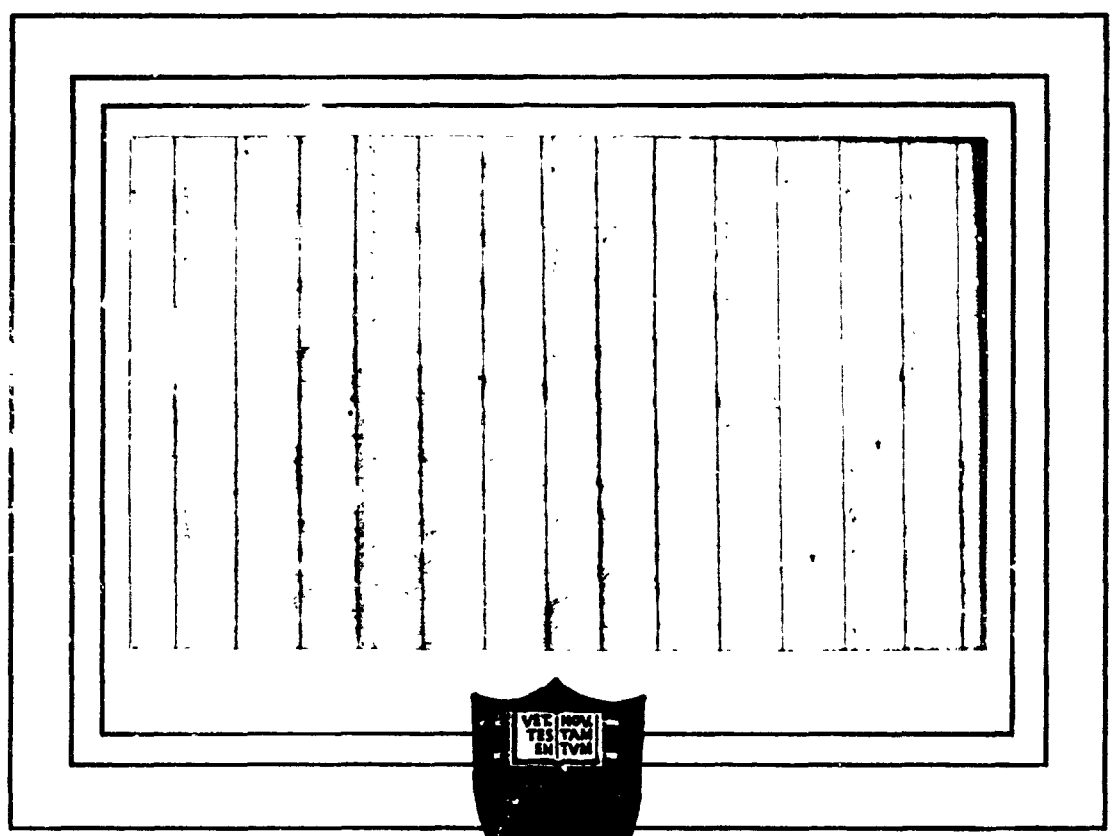


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Fluid Mechanics of Continuous Flow Electrophoresis

Final Report

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ABSTRACT

The following aspects of continuous flow electrophoresis were studied: flow and temperature fields, hydrodynamic stability, separation efficiency, and characteristics of wide-gap chambers (the SPAR apparatus). Simplified mathematical models were developed so as to furnish a basis for understanding the phenomena and comparison of different chambers and operating conditions. Studies of the hydrodynamic stability disclosed that a "wide-gap" chamber may be particularly sensitive to axial temperature variations which could be due to uneven heating or cooling. The mathematical model of the separation process includes effects due to the axial velocity, electro-osmotic cross-flow and electrophoretic migration, all including the effects of temperature dependent properties.

INTRODUCTION

Hydrodynamics plays varied roles in the continuous flow electrophoresis of small particles, in some situations the suspending fluid does little more than carry particles through the apparatus, in others the flow is so convoluted that electrophoretic separation is impossible. One of the complicating factors is the role of buoyancy forces which can destabilize the flow or establish an unfavorable, but steady laminar flow. To circumvent such problems it has been suggested that the apparatus be operated in a microgravity environment where, due to the reduced size of buoyancy forces, the chamber could be made larger and field strength increased. Then populations of large biological particles could be fractionated into narrow subpopulations on the basis of unique surface characteristics which are reflected in the electrophoretic mobility. Such an undertaking obviously requires careful evaluations of many types. The purpose of this investigation is to furnish a basis for understanding the hydrodynamic characteristics of the chamber and their effects on the separation process. Particular emphasis is placed on the role buoyancy plays in establishing the basic flow and affecting its stability.

Work began on this project in February of 1977 with the objective of assembling and evaluating current knowledge of the hydrodynamics of continuous flow electrophoresis. Four tasks were specified for the one year contract period:

- (1) Develop models to describe the flow and temperature fields;
- (2) Investigate the hydrodynamic stability of the flow field;
- (3) Develop a model to predict electrophoretic separation efficiency;
- (4) Review the SPAR apparatus and experiment.

Work on these tasks is complete insofar as it is covered by this contract and results are described in this report. The studies begun here continue under a separate NASA contract with Princeton University. The main part of the report is divided into two parts: SUMMARIES AND CONCLUSIONS, and DESCRIPTION OF RESULTS, where more detailed information is set forth.

SUMMARIES AND CONCLUSIONS

More detailed information on the various subjects is contained in the DESCRIPTION OF RESULTS sections, here we simply summarize and discuss conclusions.

Flow and Temperature Fields

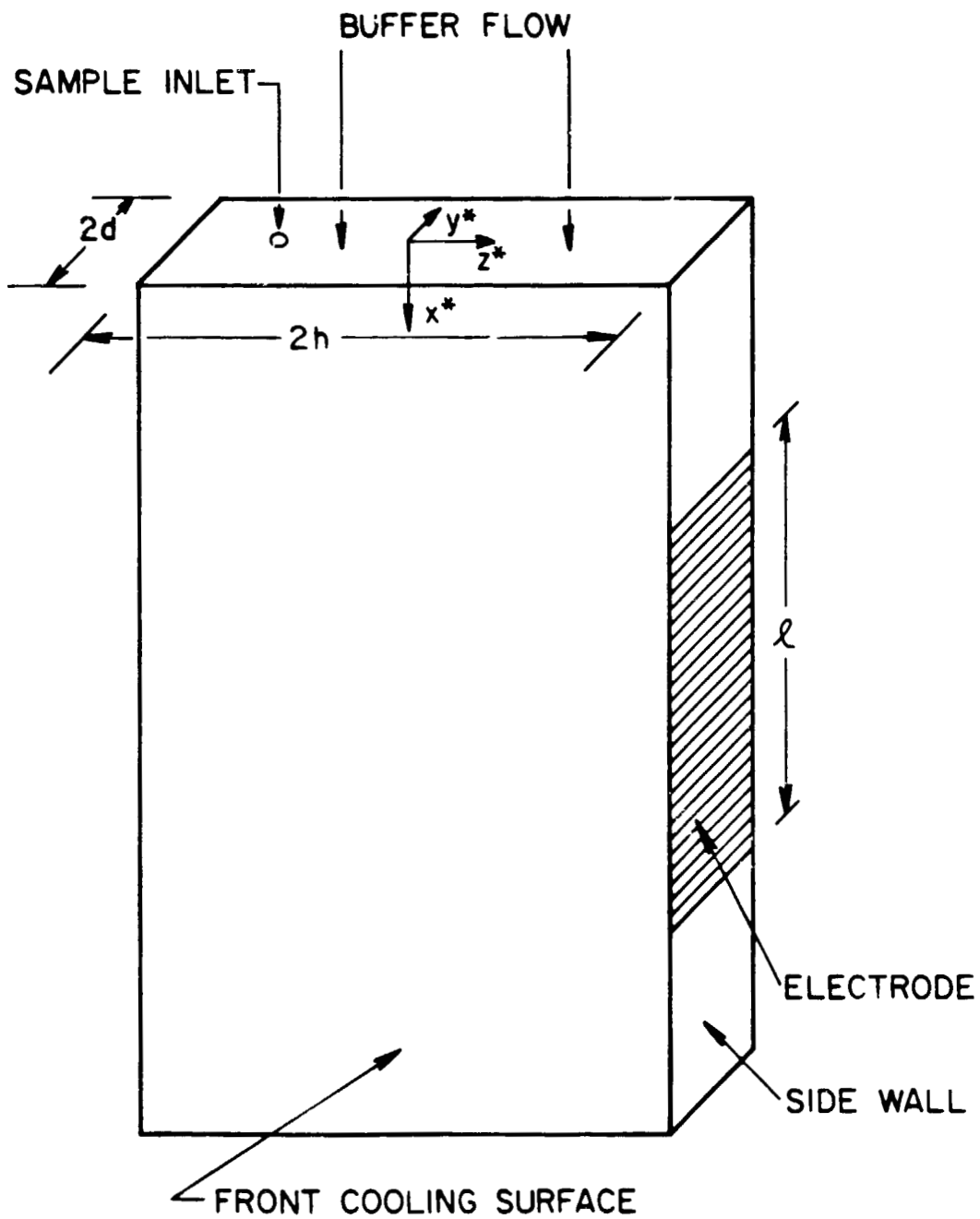
The temperature field enters the problem because it alters the electrophoretic mobility of the particles and causes density contrasts which lead to buoyancy driven flows. The non-uniform temperature field itself derives from heat effects associated with the electric field and current. Although the field is three-dimensional, it is possible to simplify matters using perturbation methods. For present purposes we sought to establish the edge effects due to cooling through the side walls containing the electrodes (cf. Figure 1)^{*}, the effects of temperature dependent conductivities for heat and electricity, and estimate time scales for thermal equilibration of the chamber.

The edge effects were found to be substantial in that they extend into the chamber for distances of 1-2 chamber thicknesses from each side. This alters the mobility of particles in these regions and has a dramatic effect on the flow through buoyancy effects. It was also found that the effects of temperature dependent conductivities were substantial, the calculated temperature rise being 70% larger for a wide-gap chamber (0.5 cm thick)^{*} than that calculated assuming constant properties. Although temperature relaxation times for the fluid are only 10-15 seconds for a narrow-gap chamber (0.15 cm) 2-3 minutes are required to establish the steady field in a wide-gap chamber.

* Representative dimensions, fluid properties, operating conditions, etc., are summarized on Table I, p. 72. A schematic diagram is on p. 7.

FIGURE 1.

Schematic representation of an electrophoresis chamber.



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For the flow field analytical solutions in two-dimensions were constructed to investigate ways buoyancy could alter the axial flow and to study edge effects. One-dimensional models were then developed to investigate the effects of temperature dependent transport properties on the axial flow and electro-osmotic cross flow.

Several conclusions can be drawn from this part of the study.

(a) It is necessary to include effects of temperature on transport properties. Models which ignore this, or treat matters inconsistently, can be qualitatively and quantitatively misleading, especially with wide-gap machines like the SPAR device. With narrow-gap machines operated with modest field strengths (cf. Table 1) the use of 'average' values is satisfactory since temperature variations are usually small.

(b) In wide-gap machines operating in a 1-g environment the steady-state axial velocity profile is unsatisfactory at modest field strengths insofar as electrophoretic separations are concerned. An earlier study by Ostrach (using a 'constant properties' one-dimensional model) identified a buoyancy-driven feature which made downflow operation unsatisfactory. Edge effects and alterations due to temperature depending properties accentuate the buoyancy feature making matters worse. Upflow, which was once suggested as a means of overcoming the difficulty, turns out to be only marginally better for the cases studied. In downflow the difficulty arises from a recirculating eddy in the center of the chamber; in upflow two eddys appear, attached to the front and rear cooling surfaces, and these restrict the area available for separation. A micro-gravity environment would suppress or eliminate secondary flows of this sort. Of course other means of eddy suppression ought not be ruled out.

(c) Experiments at General Electric using the wide-gap SPAR apparatus disclosed a meandering sample flow pattern thought to be evidence of the structure noted in (b). Subsequent calculations made with the models developed here showed that the actual power levels were far lower than those required according to the theory and thus the meandering flow must be due to another process.

Hydrodynamic Stability

In an attempt to ascertain the cause of the meandering observed in the General Electric experiments the stability of several chamber configurations was examined. Attention focussed on buoyancy driven instabilities for obvious reasons and investigations of other sorts of instability, e.g., those due to viscosity stratification or electrokinetic effects, etc., were deferred. Three sorts of instability were investigated: the inception of cellular motion due to heat generation in a quiescent layer, roll cells in a buoyancy driven shear flow and the effect of an axial temperature gradient on a fully developed flow. Critical temperature differences for the quiescent layer or the shear flow are much larger than those present in the experiments. For the vertical chamber with axial flow a new two-dimensional instability was identified with an especially low critical Rayleigh number. For conditions characteristic of the SPAR machine the critical axial gradient is (ca.) $0.5^{\circ}\text{C}/\text{cm}$.

Further experimental work will be required to establish whether or not the flow 'meandering' is a manifestation of the instability predicted by the current theory. If it is, then a micro-gravity environment will provide a means of avoiding it. Other types of instability mechanisms should also be investigated, however, so as to provide a comprehensive picture.

Prediction of Electrophoretic Separator Performance

Using the flow and temperature fields described earlier, a mathematical model of continuous electrophoretic separation was developed. The model (in brief):

(a) Accepts as input data the dimensions of the chamber, operating conditions and flowrates, transport properties of the buffer, location and size of the sample injection tube, mobility distribution of the sample, zeta-potential of the wall coating, number and size of the sample outlet streams, etc.

(b) Predicts the mobility distribution in each of the sample withdrawal streams.

Calculations were carried out using computer programs which have been tested on "model systems". Further refinements will be made under the current NASA contract with Princeton.

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SPAR Electrophoresis Experiment

Throughout the course of this investigation attention focussed on understanding the behavior of wide-gap machines and predicting their performance. We now have models of the flow and temperature fields and can estimate the electrophoretic separation characteristics of a given device.

Although refinements will be necessary, the requisite 'first generation' models now exist for interpreting results from SPAR (or other) experiments. Furthermore, the effects of changes in process variables can be examined so as to optimize the separation.

Final Comments

Work begun during this program is being carried on under a joint program coordinated by Dr. R.S. Snyder of MSFC. These tasks include:

(a) Experimental studies (at MSFC) to ascertain the reason for flow meandering in wide-gap machines at 1-g. This will serve to prove or disprove the proposal that observed unsatisfactory operation is due to a buoyancy driven instability and assist in developing ways to circumvent the problem.

(b) Case-studies with the flow and separation models developed here to ascertain the ultimate (theoretical) capabilities of continuous flow devices.

(c) Theoretical work to extend the capabilities of the model, to develop an understanding of three-dimensional effects and finite sample concentration, and to investigate other hydrodynamic instabilities which could limit resolution. Establishing the limitations due to gravity and those arising from other phenomena.

Frequent observations of particle agglomeration and 'clumping' phenomena with cells underscore the need for an investigation of effects due to particle concentration. Recent theoretical studies (Batchelor, 1972) suggest substantial changes in sedimentation velocity at particle concentrations of a few percent but none of the extant studies deal with electrokinetic effects present in electrophoresis.

(d) The development of experimental techniques to test the model using mixtures of well-characterized particles. This will include micro-gravity experiments where appropriate

DESCRIPTION OF RESULTS

I. Flow and Temperature Fields

Introduction

Inside a continuous flow electrophoresis chamber of the sort depicted on Figure 1 the temperature and velocity have a three-dimensional character. Cold buffer enters one end of the chamber and adjusts to the new geometry within a distance, x_e , which is given roughly by the formula (Schlichting, 1960)

$$x_e = 0.16 \text{ Re } d. \quad (1)$$

Here d stands for the half-thickness and Re for the Reynolds number, $u_0 d / \nu_0$; u_0 is the mean axial velocity and ν_0 the kinematic viscosity. Since the Reynolds number lies in the range 1-5, the entrance length is relatively short and here the velocity field can be modelled as being fully developed (viz. independent of x). As the buffer flow moves into the electrode region heat is added (volumetrically) through the action of the electric field and the associated current, so, to limit the temperature rise, the front and back walls are kept cold. For reasons that will be explained later (in the section on stability) the adjustment length for the temperature field can be substantial. Thus, in the electrode region the three-dimensional nature of the temperature field alters the structure of the velocity field through its effects on density and viscosity. Another contributory factor is the electro-osmotic cross-flow caused by the action of the field on the thin layer of charge in the fluid adjacent to the lateral boundaries. Although this velocity is typically much smaller than the axial velocity, it has a major role in altering the electrophoretic separation processes.

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The models developed here to describe the temperature and velocity fields take advantage of three facts:

- (i) The magnitude of the electro-osmotic velocity, w_0 , is small compared to u_0 .
- (ii) The axial variation of the temperature is slow.
- (iii) The effect of temperature on thermal conductivity and electrical conductivity is approximately linear over the temperature range of interest: $0^\circ - 35^\circ\text{C}$.

These facts justify the use of perturbation methods to develop a description of the temperature and velocity fields. First, because of (i), the velocity field can be split into two parts, an axial flow field due to forced and natural convection with a superimposed electro-osmotic flow. Next, due to the slow variation of the transport properties with axial position, (ii), the velocity fields can be split into a fully-developed part (independent of x) with corrections added later to allow for axial structure. Finally, the simple linear variation, (iii), makes the description of the temperature field particularly simple.

Separate parts of the sequel are devoted to:

- A. Mathematical models for the structure of the temperature field
 - a. A two-dimensional model in which the effects of temperature on thermal conductivity are suppressed. This provides a means of evaluating edge-effects due to heat transfer near the side wall electrodes.
 - b. A one-dimensional model to evaluate effects of a temperature dependent thermal conductivity.
 - c. A transient heat conduction model to estimate thermal relaxation times.

- B. Models for the structure of the axial velocity field
- a. A two-dimensional, constant properties model provides a means of examining edge effects and buoyancy effects using the Boussinesq approximation.
 - b. A one-dimensional, variable properties model is the basis for evaluating thermal effects and is used in the separation model described in Part III.
- C. Models for the electro-osmotic cross-flow velocity
- a. A one-dimensional, variable properties model provides a basis for evaluating thermal effects and is used in the separation model (Part III).

Temperature Field

The equation for the conservation of thermal energy is

$$C_p u \frac{\partial T}{\partial x} = \nabla \cdot k \nabla T + \sigma_0 E_0^2 \quad (2)$$

The symbols are: C_p - volumetric heat capacity, k - thermal conductivity, σ - electrical conductivity, and E_0 - the electric field strength. E_0 is assumed to be a constant throughout the analysis. Both k and σ vary with temperature in a linear fashion (see Figure 2) and so we write

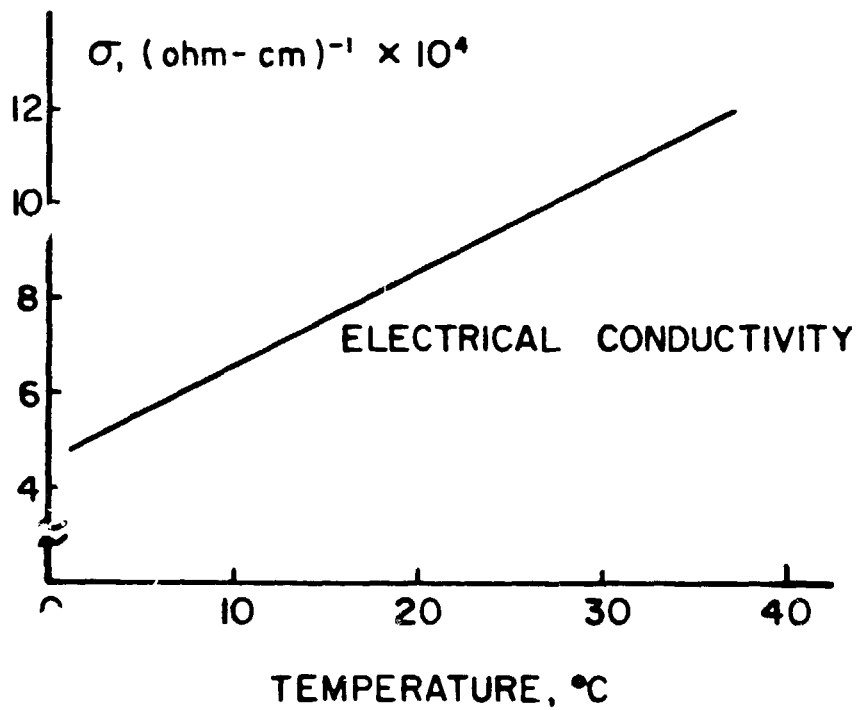
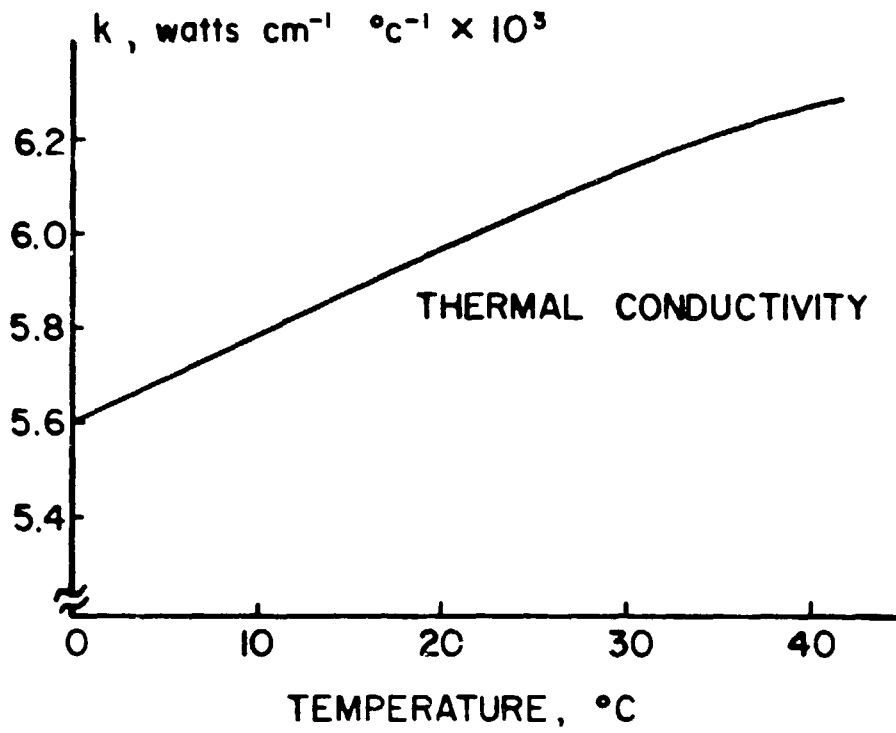
$$\sigma = \sigma_0 (1 + \sigma_1 \theta) \quad (3)$$

$$k = k_0 (1 + k_1 \theta) \quad (4)$$

where k_0 and σ_0 are reference values evaluated at the wall temperature and θ stands for a dimensionless temperature, i.e.,

FIGURE 2.

Thermal conductivity and electrical conductivity of the A-1 buffer.



$$T = T_w + \Delta T \theta \quad (5)$$

The wall temperature is T_w and ΔT is a characteristic temperature difference. If we transform to dimensionless variables with u_0 as the characteristic velocity and d the characteristic length, then

$$Pe u \frac{\partial \theta}{\partial x} = \nabla \cdot [(1 + k_1 \theta) \nabla \theta] + 1 + \sigma_1 \theta \quad (6)$$

where $Pe = C_p u_0 d / k_0$ and $\Delta T = \sigma_0 E_0^2 d^2 / k_0$.

Both k_1 and σ_1 are less than unity: k_1 is typically $O(10^{-2})$ while σ_1 is $O(10^{-1})$, and so it is convenient to represent the temperature by means of a perturbation series in k_1 , viz.

$$\theta = \theta^{(0)} + k_1 \theta^{(1)} + k_1^2 \theta^{(2)} + \dots \quad (7)$$

Substituting into (6) we generate a sequence of equations for $\theta^{(0)}$, $\theta^{(1)}$, ...

$$Pe u \frac{\partial \theta^{(0)}}{\partial x} = \nabla^2 \theta^{(0)} + \sigma_1 \theta^{(0)} + 1 \quad (8)$$

$$Pe u \frac{\partial \theta^{(1)}}{\partial x} = \nabla^2 \theta^{(1)} + \sigma_1 \theta^{(1)} + \nabla \cdot (\theta^{(0)} \nabla \theta^{(0)}) \quad (9)$$

etc.

From equation (8) we deduce that in the thermal entrance region $\partial \theta^{(0)} / \partial x$ is $O(Pe^{-1})$ and varies exponentially. Viscosity and density also depend on temperature so that a description of the temperature field to the order implied by equations (8) and (9) would entail an expansion for the axial velocity of the form

$$u = u^{(0)} + u^{(1)} + \dots \quad (10)$$

where $u^{(0)}$ is the constant properties solution, $u^{(1)}$ accounts for temperature variations, etc. This expansion would be used to furnish complete velocity fields for (8) and (9). Because of the complexity of the problem it has not been practical here to attempt a solution which includes axial variations. Instead we have suppressed the axial structure and developed a 'zero-order' approximation with which we can assess the orders-of-magnitude of the thermal effects. An investigation of the details of the axial structure of the temperature and velocity fields is part of the work being done now under another NASA contract.

Two-Dimensional, Constant Thermal Conductivity Model

The major features of the fully developed temperature field can be found by solving equations (8) and (9), omitting the convective terms. The boundary conditions are:

- (i) isothermal side walls, $\theta = 0$, at $y = \pm 1$.
- (ii) heat transfer through the side walls at $z = \pm H$ modelled in terms of a heat transfer coefficient, h , i.e.,

$$-k \frac{\partial T}{\partial z} = h_0(T - T_B).$$

T_B stands for the coolant temperature. For the zero-order field we have, in dimensionless form:

$$0 = -\nabla^2 \theta^{(0)} + \sigma_1 \theta^{(0)} + 1 \tag{11}$$
$$\theta^{(0)} = 0, \quad y = \pm 1, \quad \frac{\partial \theta^{(0)}}{\partial z} = -Bi \theta^{(0)}, \quad z = \pm H$$

The Biot number, Bi, is $h_0 d/k_0$; $H = h/d$. For $Bi = 0$ the end walls are perfectly insulating and the temperature field is one-dimensional. For $Bi \rightarrow \infty$ the end walls are isothermal with $\theta^{(0)} = 0$. Fourier transforms were used to solve equation (11) and the solution is

$$\theta^{(0)}(y,z) = \frac{2}{\pi} \sum_1^{\infty} [A_n \cosh \lambda_n z + B_n] \sin \frac{n\pi}{2} (1+y) \quad (12)$$

where

$$A_n = \frac{Bi}{\lambda_n^2} \frac{(1+(-1)^n)}{n} (\lambda_n \sinh \lambda_n H - Bi \cosh \lambda_n H)^{-1}$$

$$B_n = \frac{1-(-1)^n}{n\lambda_n^2}$$

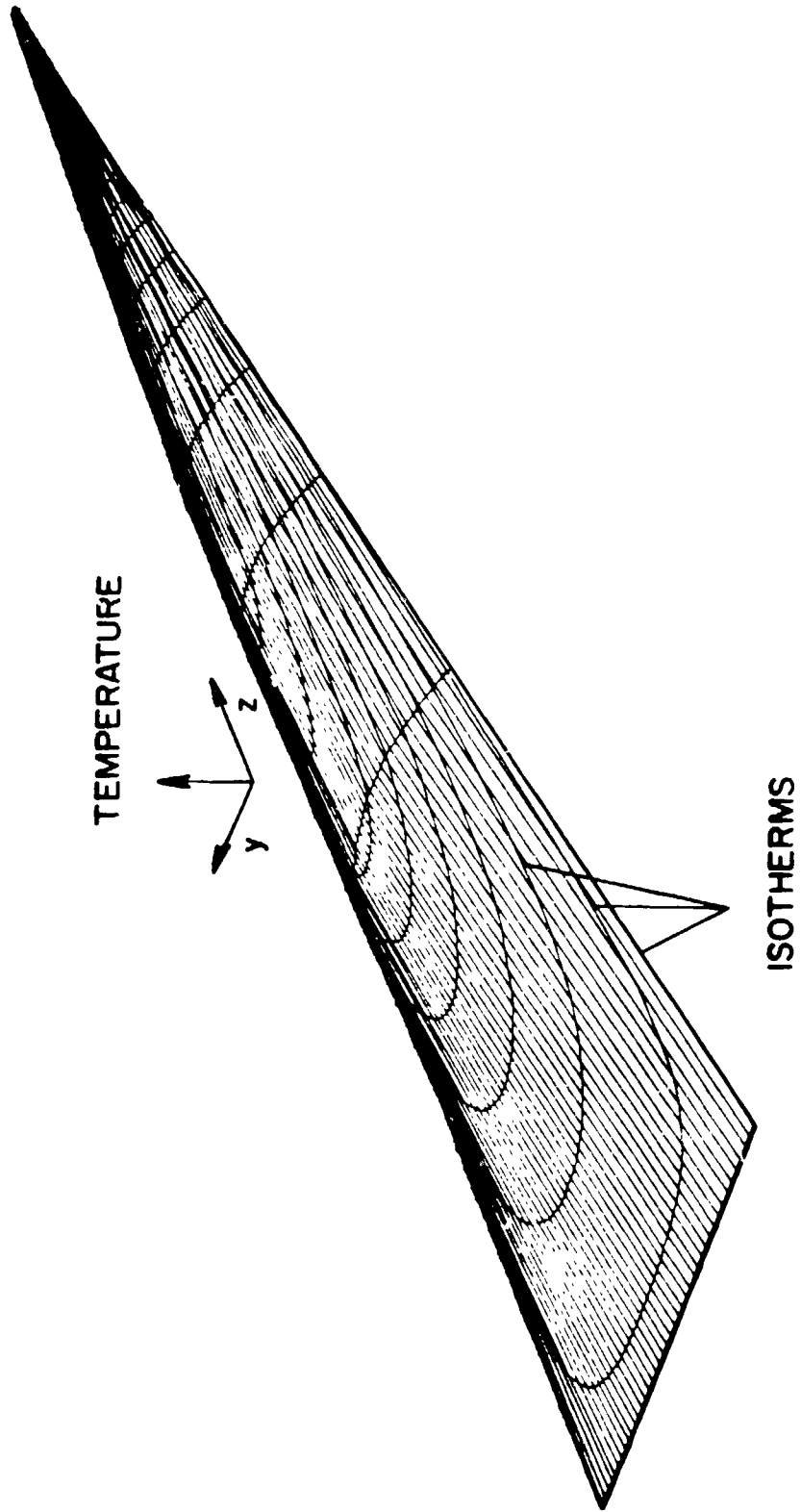
$$\lambda_n^2 = \frac{n^2 \pi^2}{4} - \sigma_1$$

Figures 3-6 display representative features of the temperature fields for narrow-gap and wide-gap chambers. Figures 3 and 5 are perspective views, Figures 4 and 6 are sections. A noteworthy feature is that the effect of side walls, shown on Figures 4 and 6, persists for a distance of 2-3 half thicknesses into the chamber at each side. For the wide-gap chamber (0.5 cm wide) this distance is (roughly) 0.75 cm and for the narrow-gap chamber (0.15 cm wide), 0.45 cm. Thus, due to the effect of temperature, particles within these regions will have a different electrophoretic mobility from those in the interior. In addition, buoyancy effects will be accentuated.

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FIGURE 3.

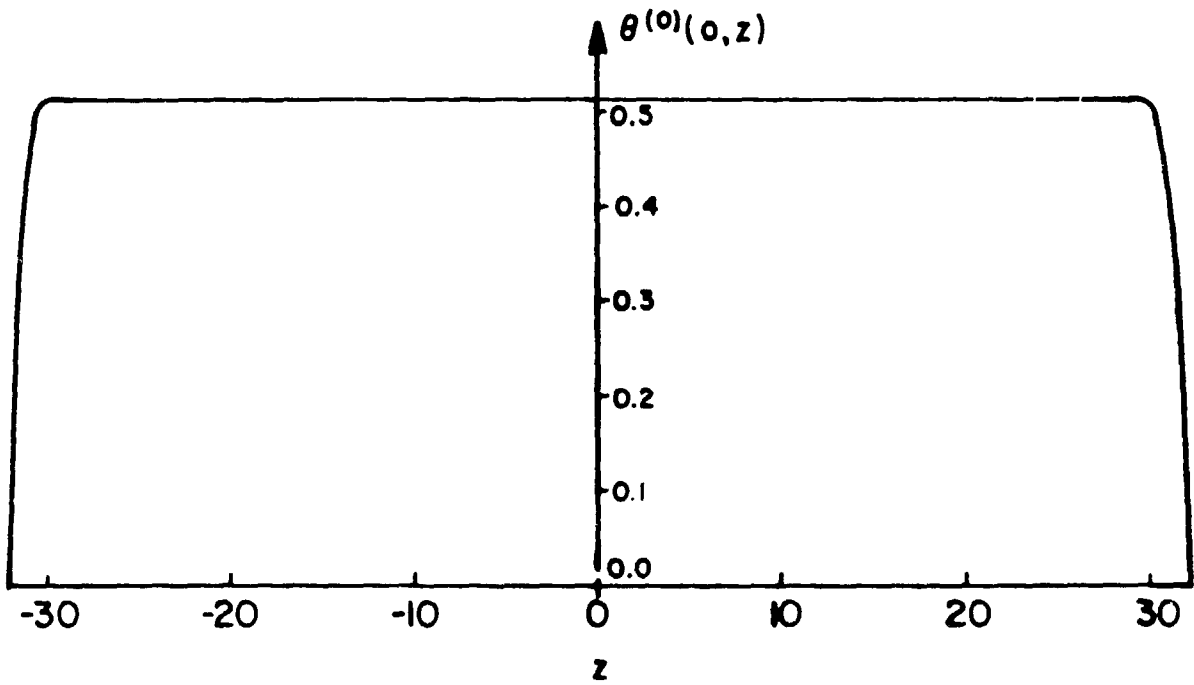
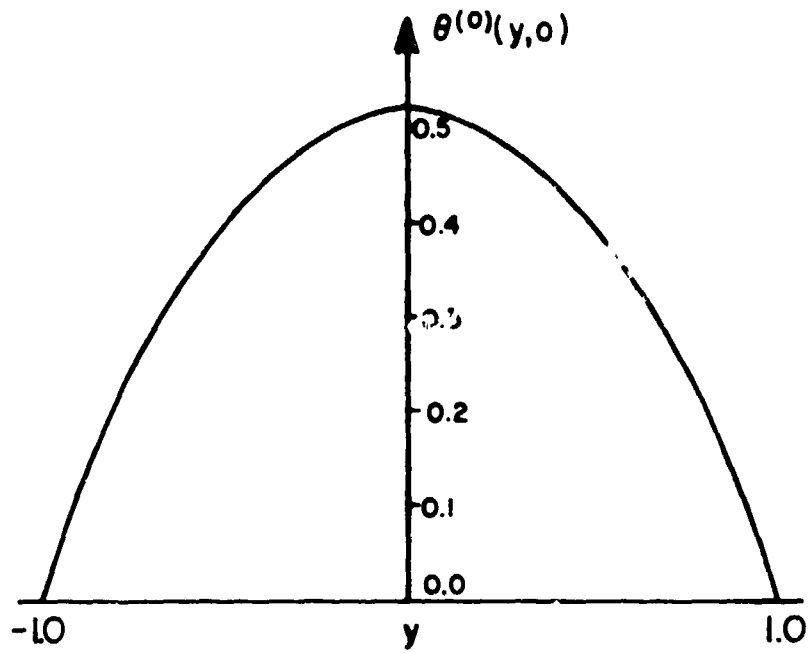
Perspective view of the temperature field for the narrow-gap chamber operating at conditions listed on Table I (uniform thermal conductivity). Note: $2d = 0.15$ cm, $2h = 5$ cm. Eq. (12).



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FIGURE 4.

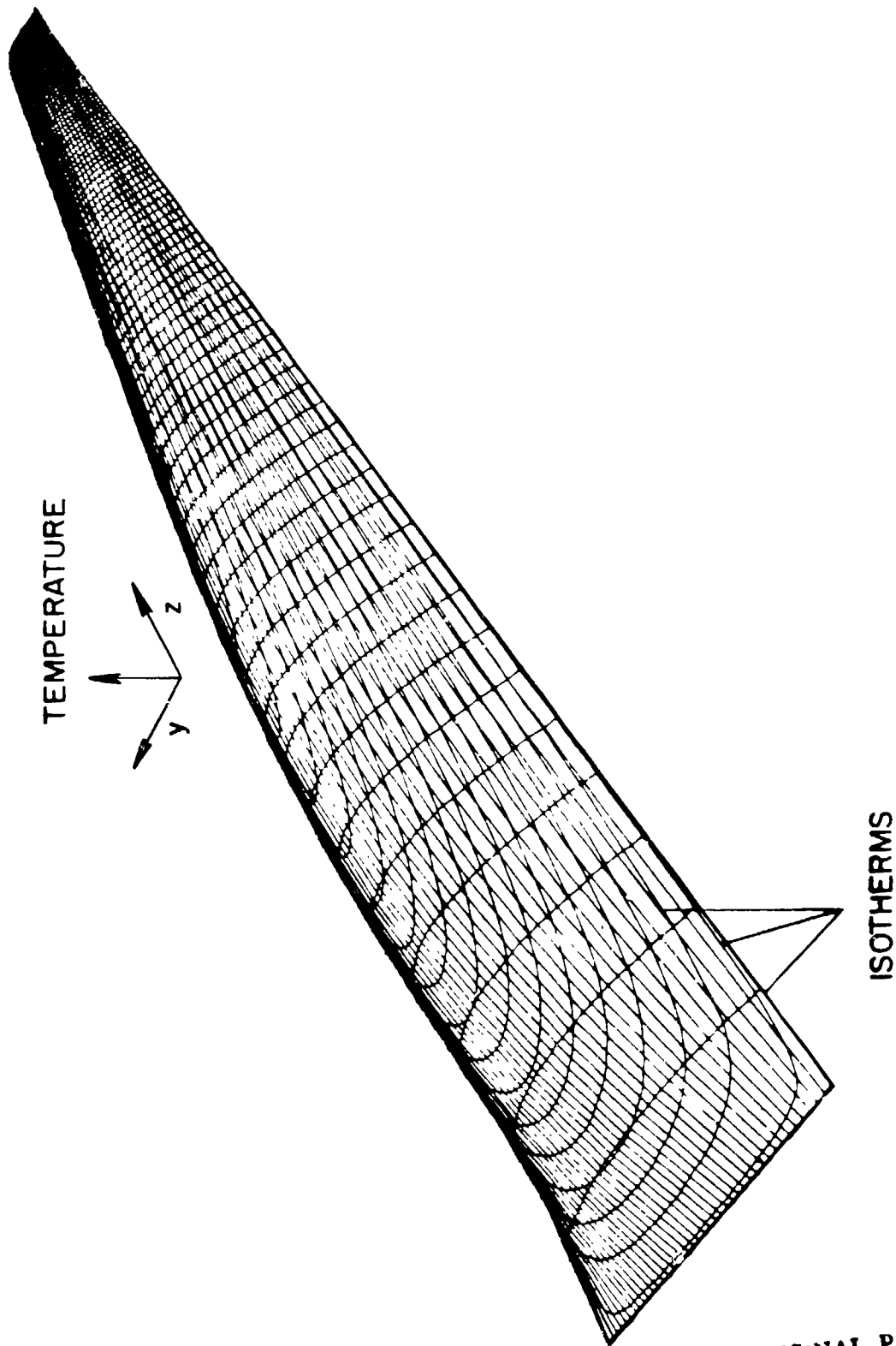
Sections of the temperature field as shown in Figure 3.



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FIGURE 5.

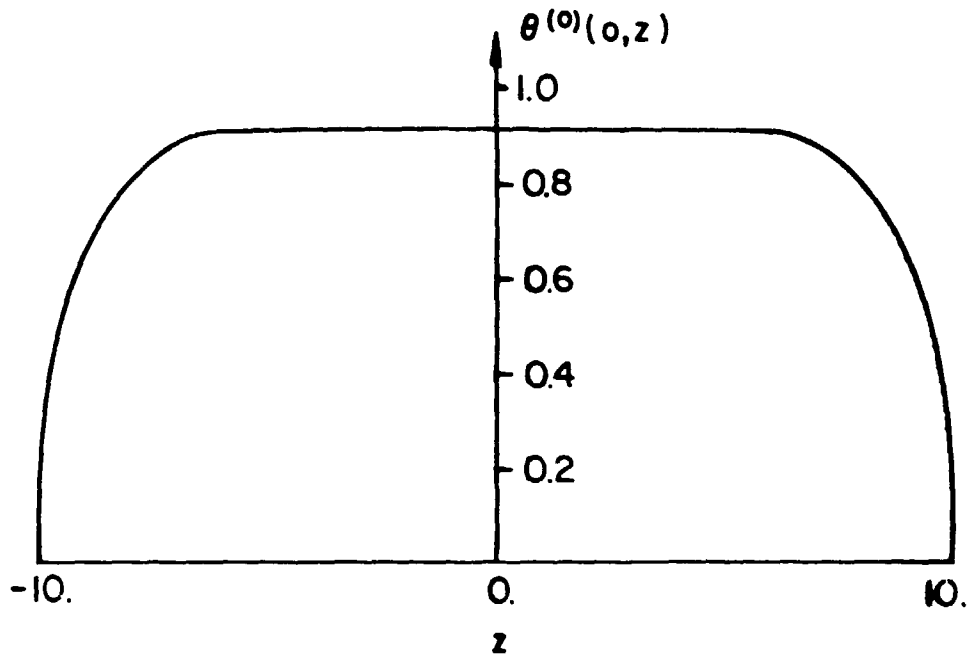
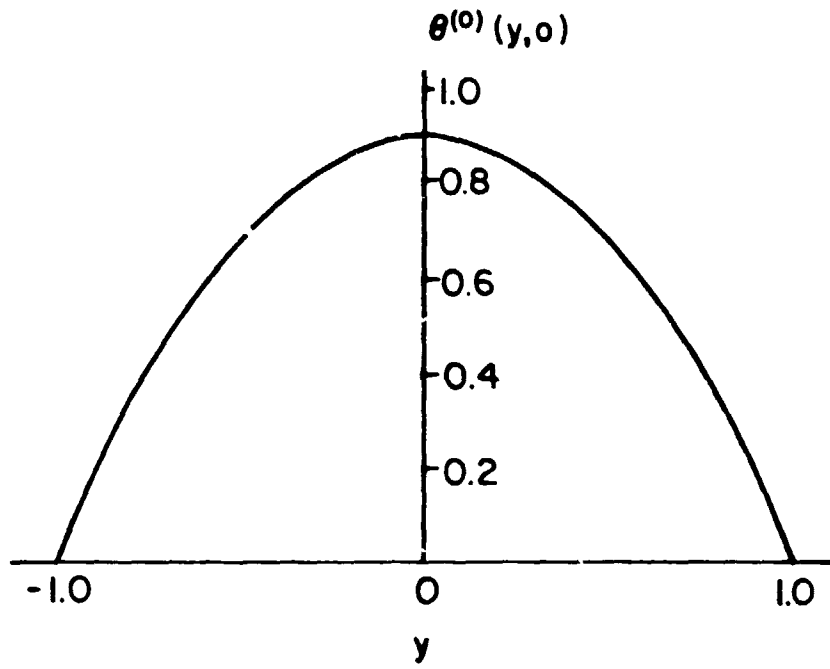
Perspective view of the temperature field for the wide-gap chamber operating at conditions listed on Table I (uniform thermal conductivity). Note: $2d = 0.5$ cm, $2h = 5$ cm. Eq. (12).



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FIGURE 6.

Sections of the temperature field as shown in Figure 5.



One-Dimensional, Variable Thermal Conductivity Model

A one-dimensional temperature field corresponds to a very wide chamber, $H \rightarrow \infty$, or one with insulated end walls at $z = \pm H$. The zero-order field is given by

$$\theta^{(0)}(y) = \frac{1}{\sigma_1} \left[\frac{\cos N_1 y}{\cos N_1} - 1 \right] \quad (13)$$

$$N_1^2 = \sigma_1$$

and perturbations are found from solutions to

$$\frac{d^2 \theta^{(1)}}{dy^2} + \sigma_1 \theta^{(1)} = -\frac{1}{2} \frac{d^2}{dy^2} [\theta^{(0)}]^2 \quad (14)$$

Using Fourier transforms the solution is found to be

$$\theta^{(1)}(y) = \frac{2}{\pi} \sum_1^{\infty} f_s(n) \sin \left[\frac{n\pi}{2} (1+y) \right] \quad (15)$$

where

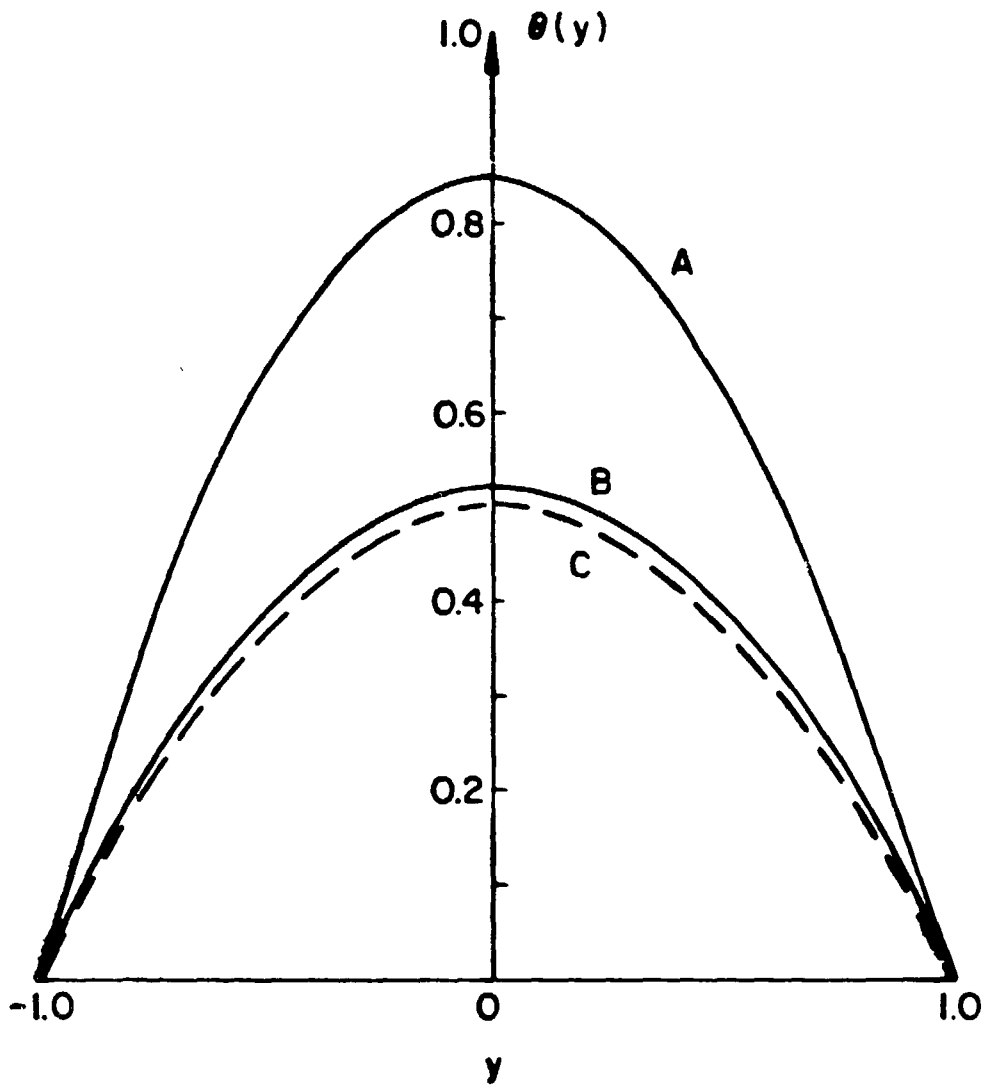
$$f_s(n) = -\frac{1}{2} \frac{n^2}{n^2 - 4\sigma_1/\pi^2} S_n\{(\theta^{(0)})^2\}$$

$$S_n\{(\theta^{(0)})^2\} = \text{sine transform of } (\theta^{(0)}(y))^2.$$

Temperature fields for narrow-gap and wide-gap chambers are shown on Figure 7, along with that for constant thermal and electrical conductivities. Figure 7 depicts matters in dimensionless form and it is seen that with the wide gap the temperature rise is 1.7 times that expected with constant properties at the conditions shown, for the A-1 buffer this is nearly 30°C.

FIGURE 7.

One-dimensional temperature field calculated so as to account for temperature dependent thermal conductivity and electrical conductivity, A - wide-gap, B - narrow-gap, C - constant properties (cf. Table I). Eqs. (13) and (15).



With the narrow-gap, the temperature rise is so small, about 1.5°C, that variable-property effects are negligible. Here we also see (upon comparison with Figure 5) that the influence of variable thermal conductivity serves to alter the temperature rise. Accounting for the variable thermal conductivity lowers this maximum by about 2°C for the wide-gap chamber at the conditions shown.

One-Dimensional, Transient Response Model

An estimate of the minimum time required to reach a steady thermal state can be made using a transient thermal model which ignores convection, since it tends to increase the equilibration time by adding colder fluid and withdrawing warm fluid from the region of interest. A rough estimate can be found from the characteristic relaxation time scale, d^2/α_0 . For the A-1 buffer $\alpha_0 = 1.39 \text{ cm}^2/\text{s}$ so that for a narrow-gap machine ($d = 0.075 \text{ cm}$) the time scale is about 4 seconds; for the wide-gap machine the time-scale is 45 seconds. To attain a condition near the steady-state generally requires 2-3 'relaxation times', as shown on Figures 8 and 9. Data for the graphs were calculated using the solution to a transient heat conduction problem with heat generation, viz.

$$\frac{\partial \theta}{\partial \tau} = \frac{\partial^2 \theta}{\partial y^2} + 1 + \sigma_1 \theta \quad (16)$$

with $\theta(\tau, -1) = 0$, $\theta(0, y) = 0$. The solution given by Carslaw and Jaeger (1959) is

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FIGURE 8.

Transient temperature response for a narrow-gap chamber (one-dimensional, constant thermal conductivity) (cf. Table I).
Eq. (17).

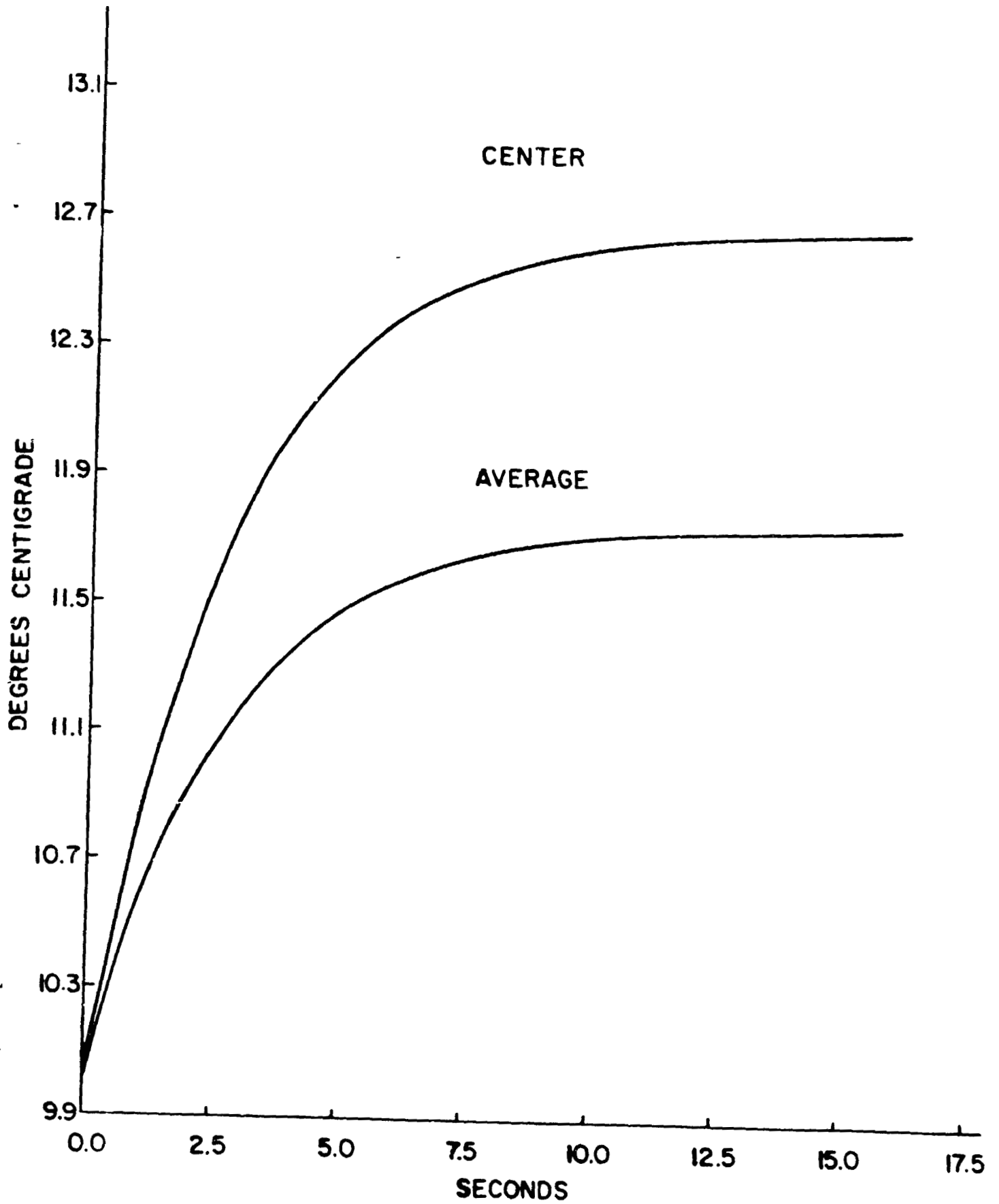
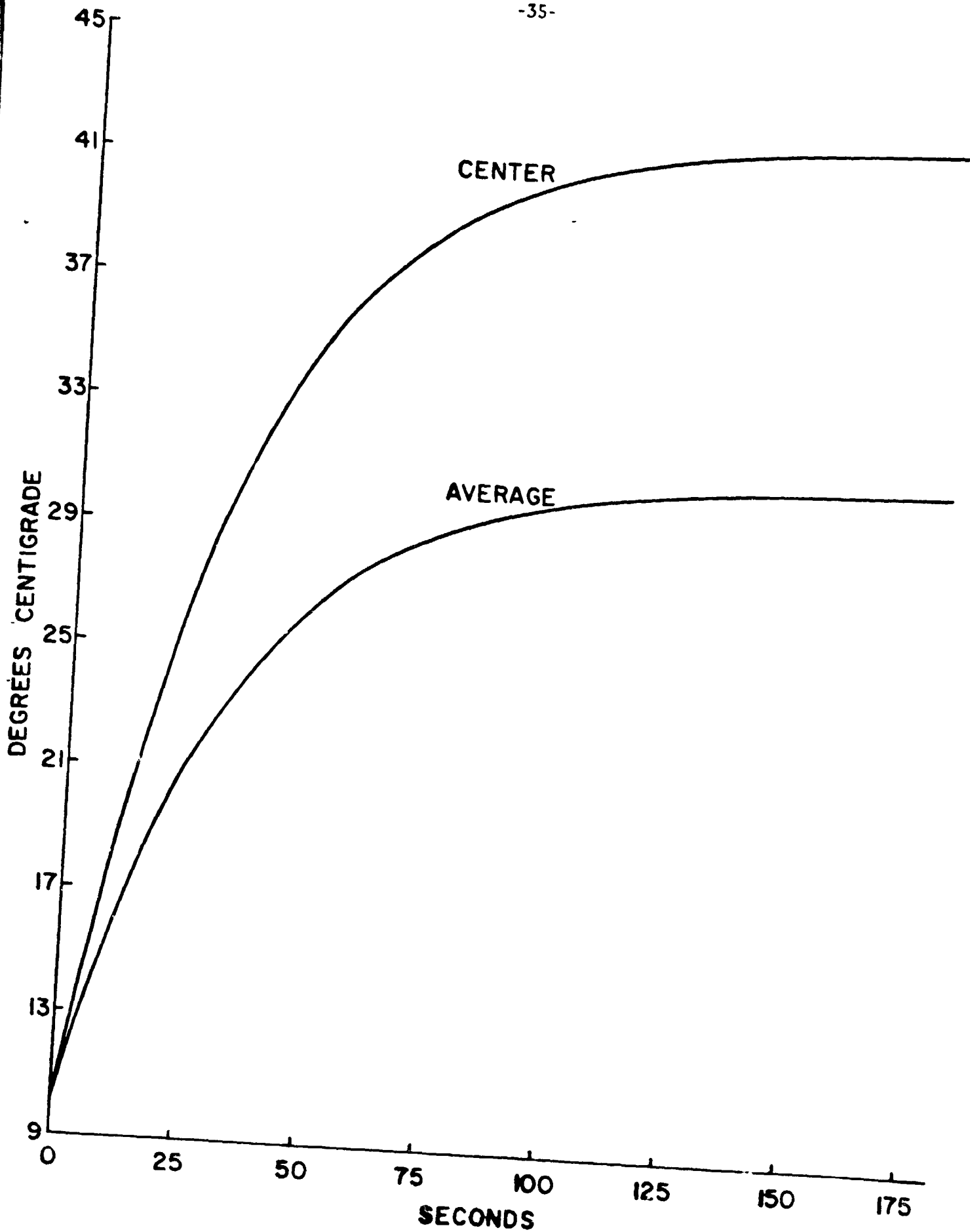


FIGURE 9.

Transient temperature response for a wide-gap chamber (one-dimensional, constant thermal conductivity) (cf. Table I). Eq. (17).



$$\theta = \frac{1}{\sigma_1} \left[\frac{\cos N_1 y}{\cos N_1} - 1 \right] + \frac{16}{\pi} \sum_0^{\infty} (-1)^n \frac{\exp(\cdot) \cos [2n+1] \pi y / 2}{[4\sigma_1 - (2n+1)^2 \pi^2] [2n+1]}$$

$$(\cdot) = [-(2n+1)^2 \pi^2 + \sigma_1] \tau \quad (17)$$

$$\tau = \alpha_0 t / 4d^2, \quad y = y^* / d$$

Obviously a more refined estimate could be made by solving a two-dimensional with convection included, however, the characteristic time for equilibration would probably not change too much. Thus, in any experiment with wide-gap machine at least 2-3 minutes should be allowed for thermal equilibration.

Axial Velocity Field

Equations to describe the axial velocity are derived from the Navier-Stokes equations using an expansion described earlier

$$u(y,z) = u^{(0)}(y,z) + u^{(1)}(y,z) + \dots \quad (10)$$

The effects of buoyancy on the two-dimensional flow field are described in the first part of this section. In the second, where the effects of the lateral boundaries at $z = \pm H$ are ignored, exact solutions are possible which account fully for the effects of temperature on viscosity and buoyancy.

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Two-Dimensional, Constant Transport Properties Model

The $0(1)$ -velocity is described by solutions to

$$0 = -K + N_2 - N_3 \theta^{(0)} + \nabla^2 u^{(0)} \quad (18)$$

where K is a constant (dimensionless) axial pressure gradient;

$N_2 = gd^3/\nu_0^2 \text{ Re}$ and $N_3 = gd^3\beta\Delta T/\nu_0^2 \text{ Re}$, $\text{Re} = du_0/\nu_0$. The parameter N_3 describes the magnitude of the buoyancy effect while $(-K + N_2)$ is a constant to be determined from the fact that the volumetric flowrate is independent of the temperature rise, since the velocity is scaled using the mean velocity, u_0 . Using Fourier Sine transforms we find

$$u^{(0)}(y,z) = \frac{2}{\pi} \sum_0^{\infty} g_s(n) \sin \frac{n\pi}{2} (1+y) \quad (19)$$

with

$$g_s(n) = B_n \cosh \frac{n\pi}{2} z + C_n \cosh \lambda_n z + D_n$$

The coefficients B_n , C_n , and D_n are found from the relations

$$B_n \cosh \frac{n\pi H}{2} + C_n \cosh \lambda_n H + D_n = 0$$

$$C_n = - \frac{N_3 A_n}{\sigma_1}$$

$$D_n = - \frac{4}{n^2 \pi^2} \left[\frac{K - N_2}{n} (1 - (-1)^n) + \frac{N_3}{n} \frac{(1 - (-1)^n)}{\lambda_n^2} \right] \quad (20)$$

Finally $(-K + N_2)$ is found from the requirement that

$$\int_0^H \int_0^1 u^{(0)}(y,z) dy dz = 4H \quad (21)$$

FIGURE 10.

Axial velocity field for the narrow-gap chamber (cf. Table I).
The velocity is almost indistinguishable from the fully developed
parabola, $3(1 - y^2)/2$ except near the side walls at $z = \pm 33.3$.
Eq. (19).

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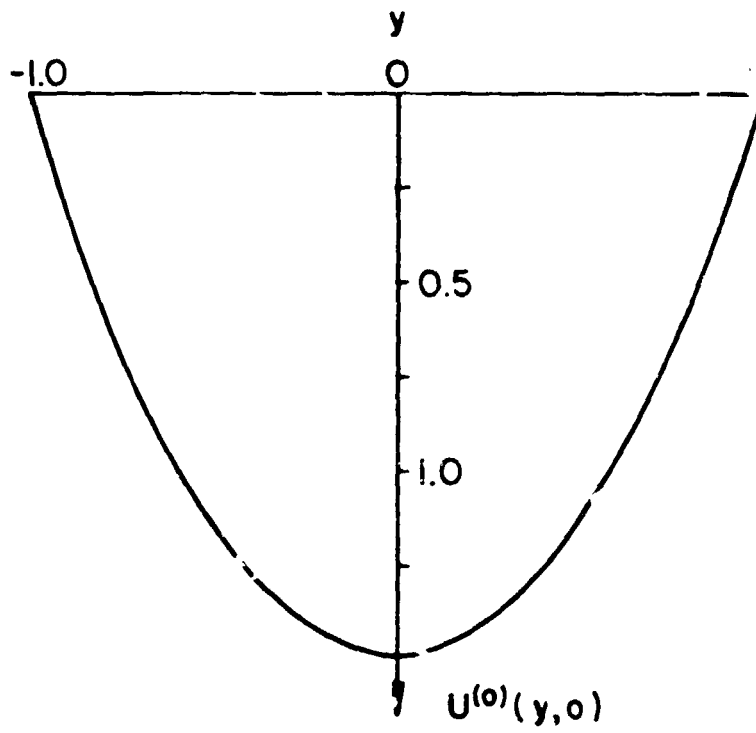
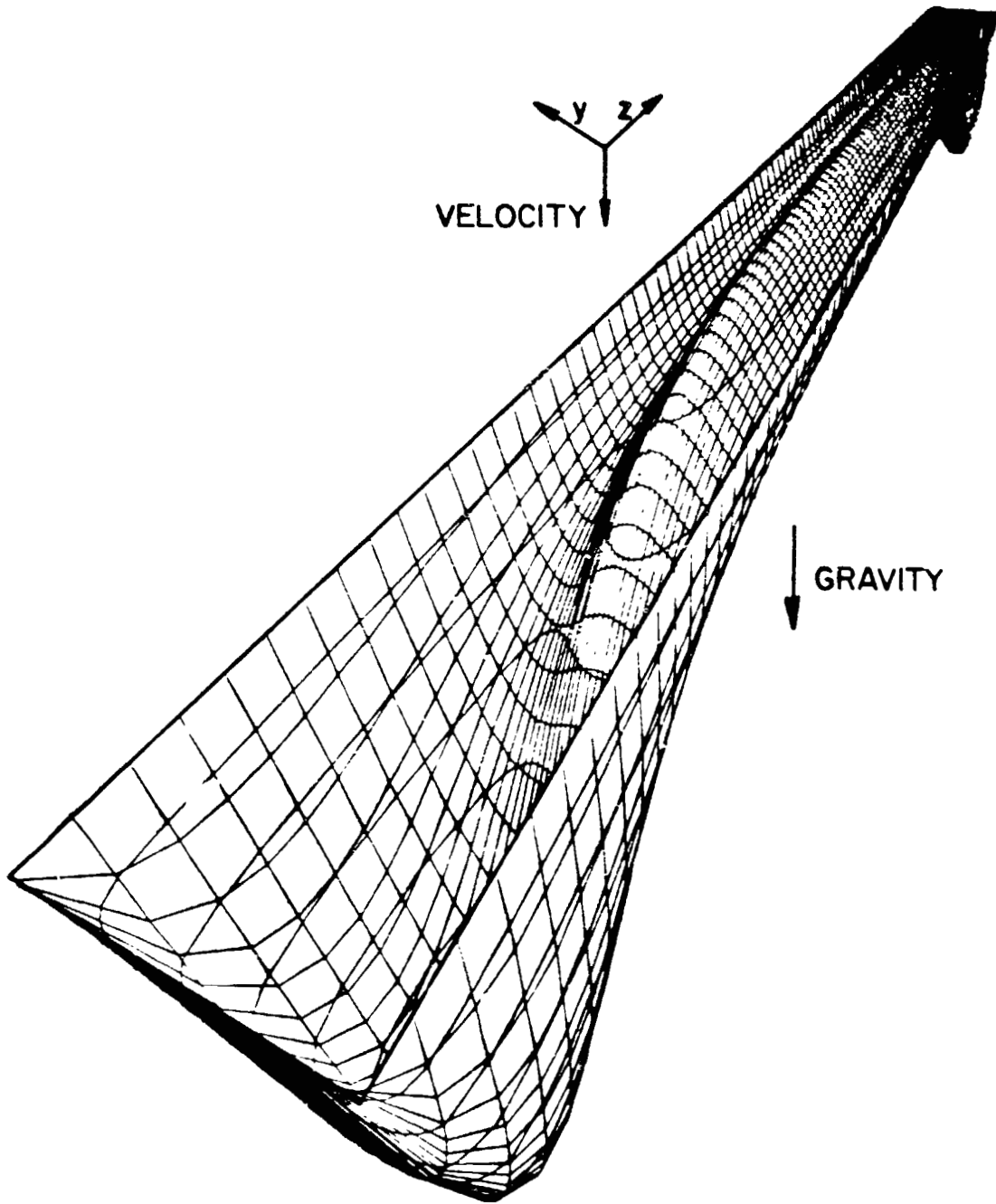


FIGURE 11.

Perspective view of the axial velocity field for the wide-gap chamber (cf. Table I). Downflow at $g = 980 \text{ cm/s}^2$. Eq. (19).

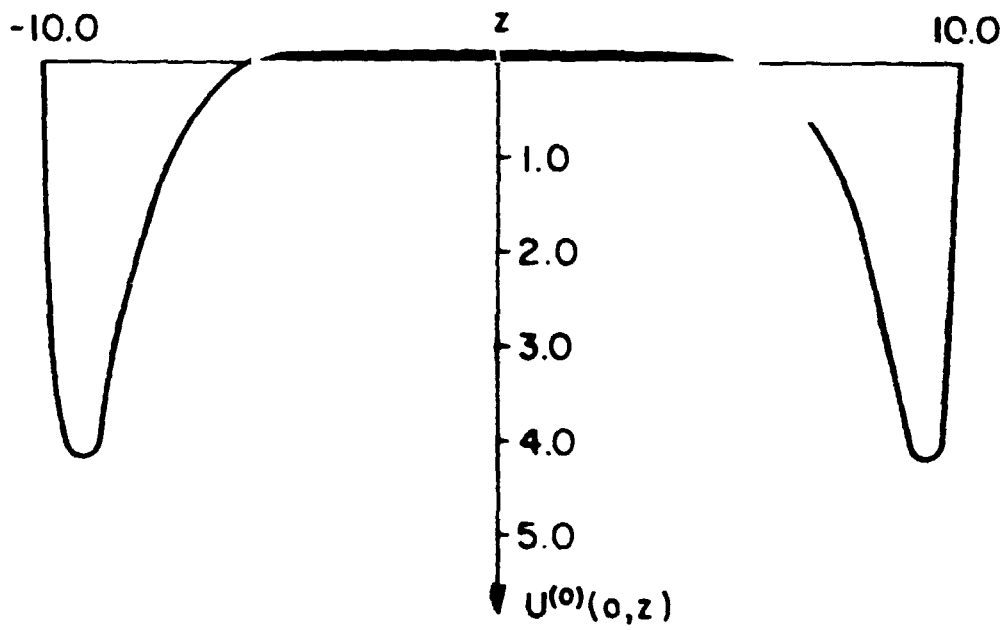
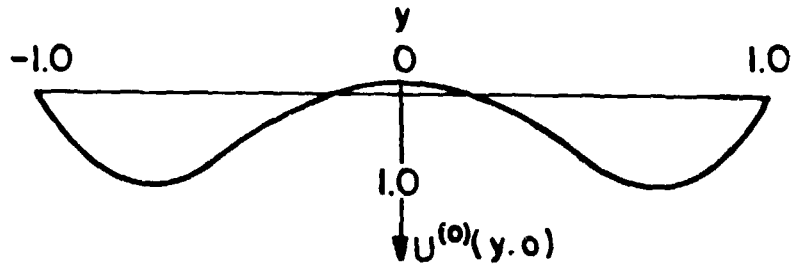
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FIGURE 12.

Central sections of the field shown on Figure 11. Note weak upflow in center and strong downflow along front and back cooling walls and near side walls at $z = \pm 10$.



↓ DIRECTION OF MAIN FLOW AND GRAVITY

FIGURE 13.

Perspective view of the axial velocity field for the wide-gap chamber
(cf. Table I). Upflow at $g = 980 \text{ cm/s}^2$. Eq. (19).

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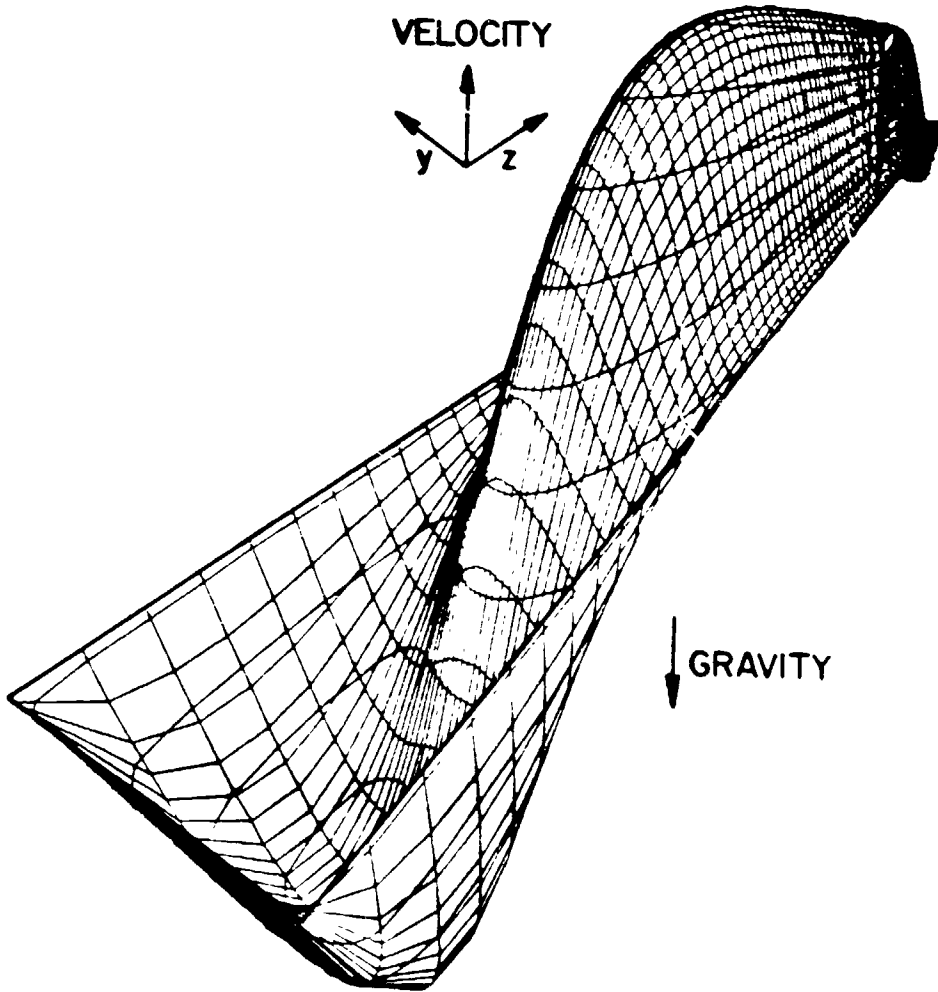
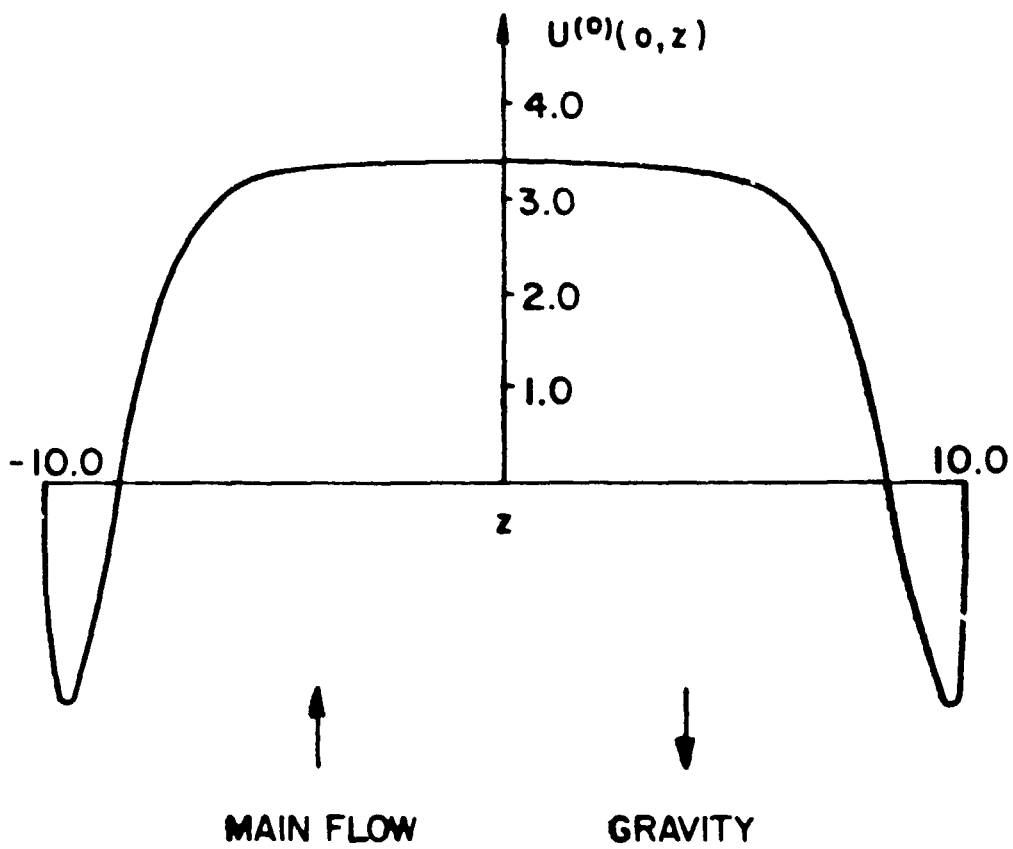
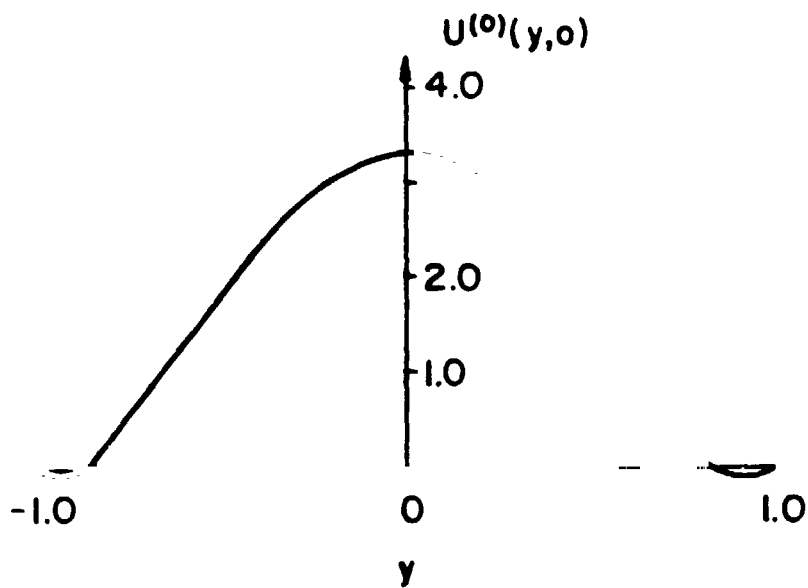


FIGURE 14.

Central sections of the field shown on Figure 13. Note downflow along front and rear cooling walls and near the side walls at $z = \pm 10$.

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Results of representative calculations are shown on Figures 10-14. Figure 10 shows that buoyancy has relatively little effect on a narrow-gap chamber due to the excellent heat transfer. The maximum temperature rise for this case is about 1°C. End effects due to the no-slip condition at $z = \pm H$ are similar to those encountered with the temperature field.

With the wide-gap chamber dramatic effects are present at unit gravity. Figure 11 is a perspective view, Figure 12 shows the velocity across two sections of the chamber with downflow. Here buoyancy causes regions of reversed flow throughout a large part of the central section. With a finite length chamber this implies that a large recirculating eddy would be present in the central section and the chamber would be difficult to use as an electrophoretic separation device. If the flow is reversed, the recirculating eddy splits apart and is attached to the side walls, as is shown on Figures 13 and 14. Since the recirculating regions are adjacent to the side walls, the central section allows more-or-less free passage of fluid and any sample. Of course, any sample that migrated into the eddy structures would be difficult to recover.

One-Dimensional Velocity Fields

Effects due to a temperature dependent viscosity and buoyancy can be investigated easily using a one-dimensional model and, in fact, exact solutions can be obtained. Since variations are confined to the y-direction we have

$$0 = K + N_2 \rho + \frac{d}{dy} \mu \frac{du}{cy} \quad (22)$$

with

$$u(\pm 1) = 0$$

The solution is

$$u(y) = -K \int_0^1 \frac{y_1}{\nu} dy_1 + N_2 \int_0^1 \int_0^{y_2} \frac{1}{\nu} \rho dy_1 dy_2 \quad (23)$$

with K determined from the fact that the velocity scale is the mean velocity. Thus

$$1 = -K \int_0^1 \int_0^1 \frac{y_1}{\nu} dy_1 dy_2 + N_2 \int_0^1 \int_0^1 \int_0^{y_2} \frac{1}{\nu} \rho dy_1 dy_2 dy \quad (24)$$

Given expressions for the dimensionless viscosity, and density, both scaled on values evaluated at the wall, it is a straightforward task to evaluate the integrals. In the calculation a two-term expression was used for the temperature, viz.

$$\theta = \theta^{(0)} + k_1 \theta^{(1)} \quad (7)$$

For purposes of comparison we can evaluate the velocity field for uniform viscosity and buoyancy. The temperature field is

$$\theta^{(0)}(y) = \frac{1}{\sigma_1} \left[\frac{\cos N_1 y}{\cos N_1} - 1 \right], \quad N_1^2 = \sigma_1 \quad (13)$$

and the velocity works out to be

$$u^{(0)}(y) = \frac{3}{2} (1-y^2) - \frac{3N_3}{\sigma_1} \left[\frac{1}{\sigma_1} + \frac{1}{3} - \frac{1}{\sigma_1 N_1} \tan N_1 \right] (1-y^2) + \frac{N_3}{\sigma_1} \left[\frac{1}{\sigma_1} + \frac{1}{2} (1-y^2) - \frac{1}{\sigma_1} \frac{\cos N_1 y}{\cos N_1} \right] \quad (25)$$

For $\sigma_1 \rightarrow 0$ we recover an earlier result due to Ostrach (1976),

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$$u^{(0)}(y) = \frac{3}{2} (1-y^2) - \frac{N_3}{120} (1-y^2)(1-5y^2) \quad (26)$$

It is important to note that the magnitude of the buoyancy effect as compared to forced convection is contained in the dimensionless group N_3 , viz.

$$\frac{g\beta}{\nu_o u_o} \frac{\sigma_o E_o^2 d^4}{k_o}$$

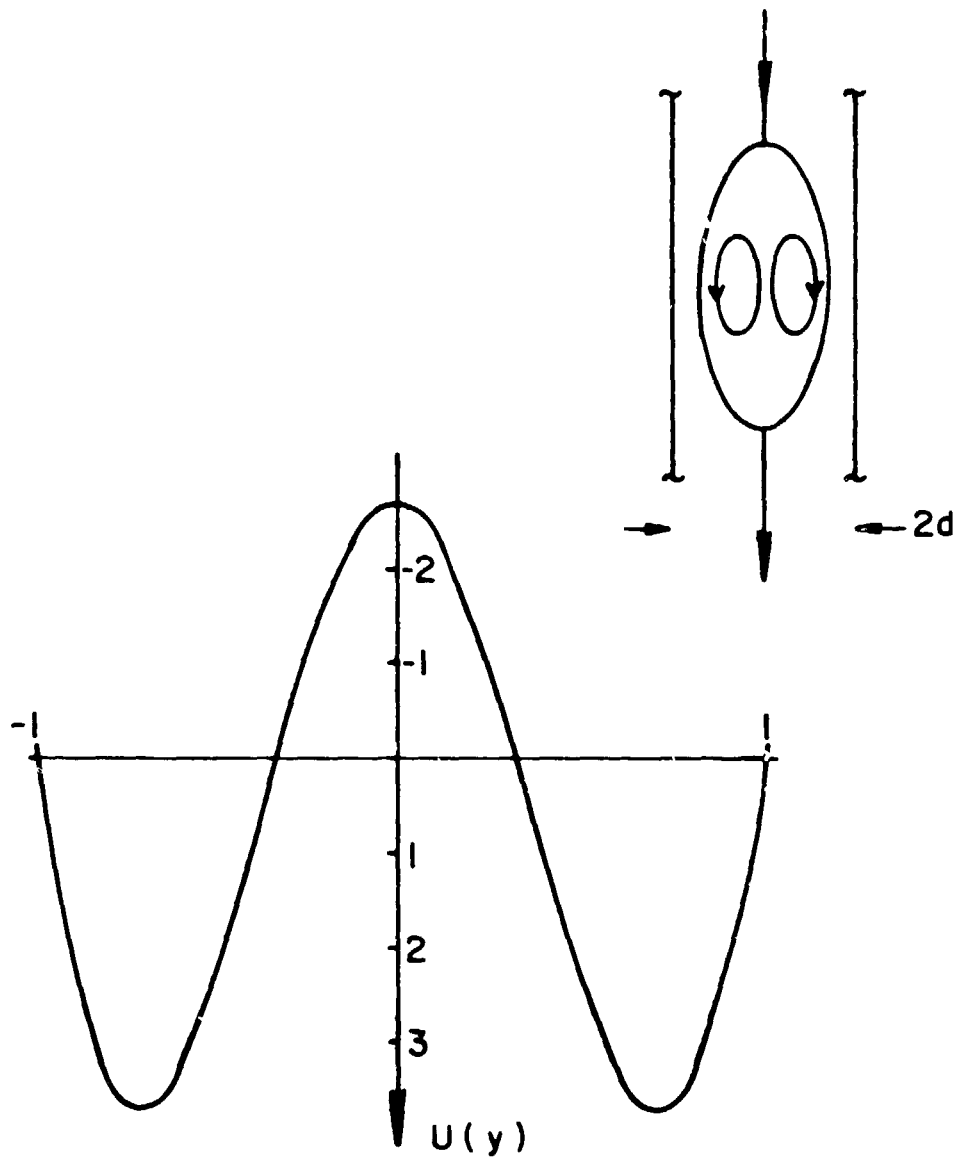
Here β represents an average coefficient of thermal expansion, $-\rho^{-1}(\partial\rho/\partial T)$. Thus, all other things being equal, going from a narrow-gap machine ($2d = 0.15$ cm) to a wide-gap machine ($2d = 0.5$ cm) alters the effect of gravity by a factor of $(10/3)^4$, roughly two orders-of-magnitude. It follows that whereas gravity forces play minor roles with narrow-gap machines the situation with wide-gap machines is quite the opposite.

For $|N_3| \ll 1$ the velocity differs very little from the familiar parabolic profile characteristic of forced convection. For $|N_3| \gg 1$ regions of reverse flow are present as illustrated on Figures 15 and 16. Qualitatively these profiles are similar to those derived from the two-dimensional model; quantitatively, however, they differ in the magnitude of the maximum velocity, which is higher here due to the effects of variable viscosity and buoyancy. Note regions of reversed flow in the center for downflow and adjacent to side walls for upflow (see insets).

The recirculating eddy present in the downflow configuration at *these operating conditions* renders this configuration almost useless for electrophoretic separation. Even if the flow was steady, the sample stream would be deflected towards the wall where electro-osmotic effects would be strong.

FIGURE 15.

One-dimensional velocity field calculated with allowance for variable viscosity and density (cf. Table I). Downflow, $g = 980 \text{ cm/s}^2$. Eq. (23).



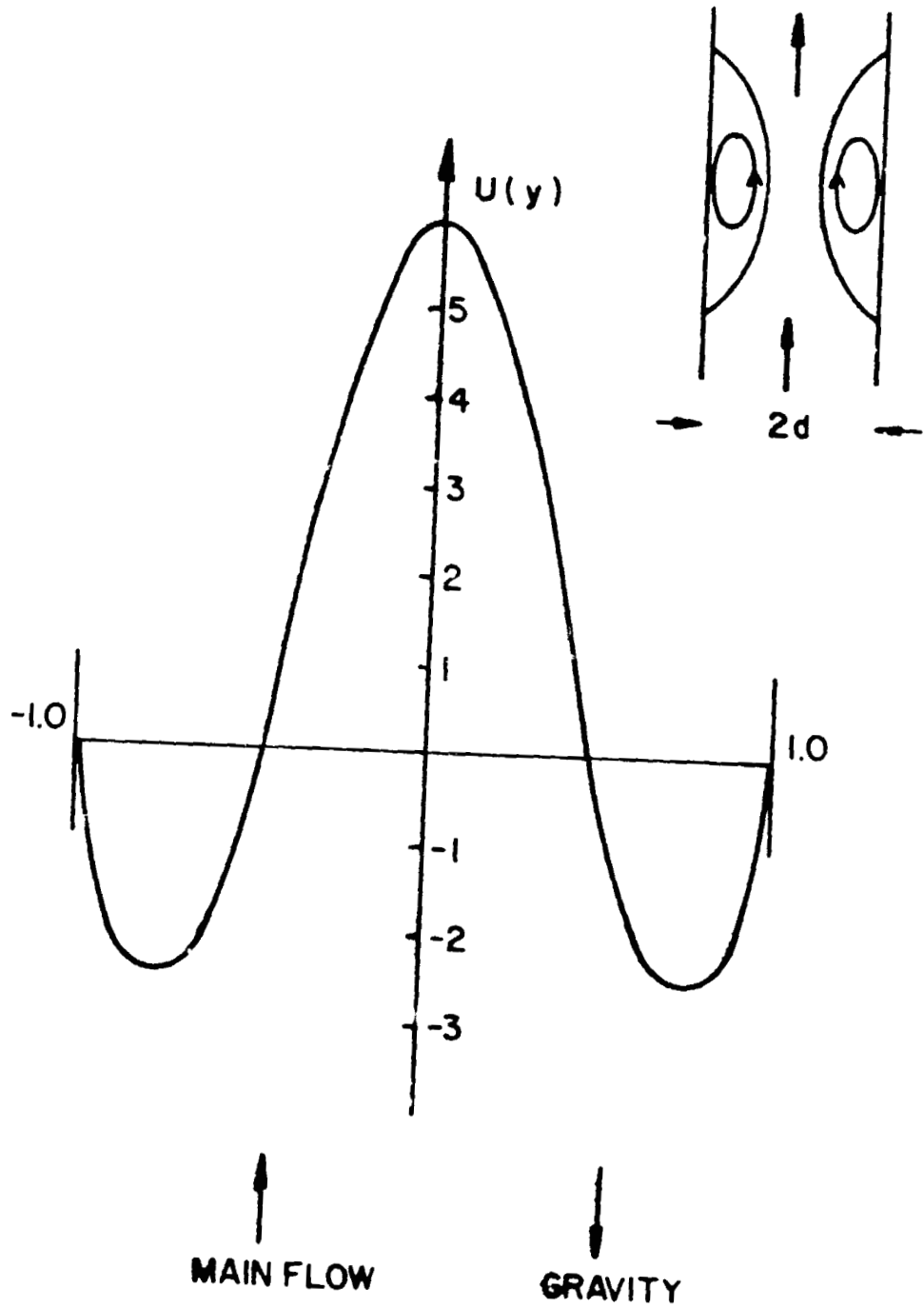
↓
MAIN FLOW

↓
GRAVITY

FIGURE 16.

One-dimensional velocity field calculated with allowance for variable viscosity and density (cf. Table I). Upflow, $g = 980 \text{ cm/s}^2$. Eq. (23).

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In the upflow configuration the eddies attached to the walls take up nearly 80% of the cross section and severely reduce the region available for sample separation.

At the same time, however, it must be recognized that the calculations presented here do not exhaust the set of configurations and operating conditions. The recirculations, for example, can be changed by independent control of the pressure gradient. Lowering the field strength has a dramatic effect on the buoyancy, since the temperature rise is proportional to E_0^2 , but would necessitate an increase in the length of the chamber, etc. There are, therefore, a number of options which remain to be investigated, the calculations given here simply illustrate the hydrodynamic phenomena.

Electro-osmotic Cross-flow Velocity Field

The presence of a thin layer of space charge adjacent to the boundaries in the y-z plane (cf. Figure 1) along with the transverse electric field causes a well-known electro-osmotic flow (Shaw, 1969). In a parallel plate system open to reservoirs at $z = \pm H$ where $H \gg 1$ the velocity profile would appear to be flat up to a very small distance from the wall (a few multiples) of the Debye thickness, κ^{-1}) where a rapid transition occurs to accommodate the no-slip condition. For most purposes the double-layer thickness, κ^{-1} is so small that we can approximate the velocity just outside this layer using one of the Smoluchowski equations,

$$w_0 = - \epsilon E_0 \zeta / \mu \quad (27)$$

where w_0 is the velocity in the z-direction, ϵ , the dielectric constant, E_0 , the field strength and, ζ , the zeta-potential of the wall material in

contact with the solution in question. This apparent slip-velocity is of the order of a few microns per second for a potential gradient of one volt per centimeter. When the flow is constrained by walls at $z = \pm H$ the profile is forced to be (roughly) parabolic so as to accommodate the condition of no net flow across the y - z plane. Although this velocity is too slow to affect the axial flow substantially the cross-flow interferes with any electro-phoretic separation by stretching the sample cross section. To provide an estimate of thermal effects and furnish a consistent representation of the velocity field for use in the separations model a simplified model is used. In this model end effects are omitted except insofar as the walls force the flow to turn round as sketched in Figure 17. The electro-osmotic velocity can be calculated from solutions to

$$0 = - \frac{\partial p}{\partial z} + \frac{\partial}{\partial y} \mu \frac{\partial w}{\partial y} \quad (28)$$

with $w = w_0$ at $y = \pm d$.

The pressure gradient arises from the need to balance viscous forces outside the double-layer so as to produce a field with no net flow. If we scale the velocity using w_0 , and lengths using d , and write the viscosity as $\mu_0 \mu(\theta)$ to account for the thermal effects we obtain

$$0 = K_2 + \frac{d}{dy} \mu \frac{dw}{dy} \quad (29)$$

which can be integrated to

$$w = K_2 \int_1^y \frac{y_1}{\mu} dy_1 - 1 \quad (30)$$

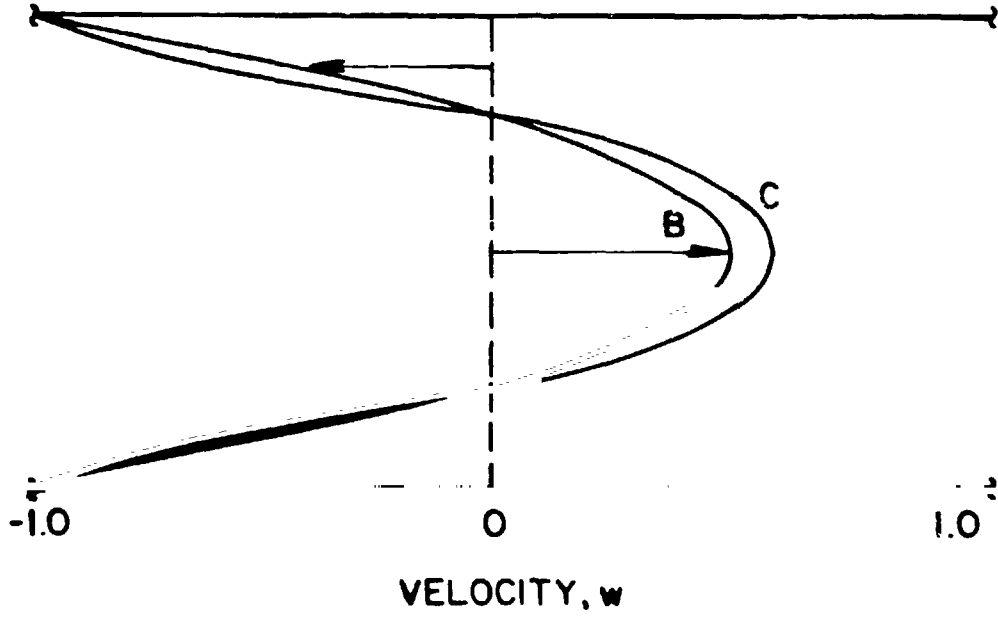
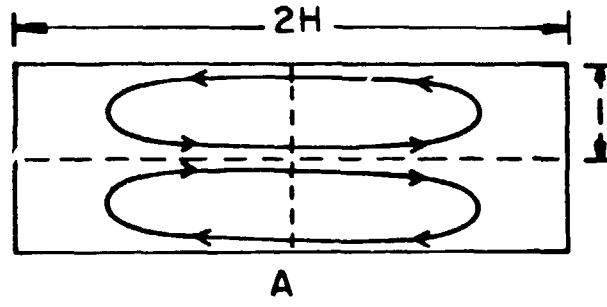
The constant k_2 is found from the requirement that the net flow across any cross section in the x-y plane be zero.

Although effects due to temperature dependent viscosity are relatively small and change the velocity only about 10% in the wide-gap chamber, these could be significant if one were modelling the separation of particles with small mobility differences.

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FIGURE 17.

Electro-osmotic crossflow velocity: A - plan view showing recirculation caused by end walls, B - velocity profile with constant viscosity, $(1 - 3y^2)/2$; C - velocity profile with wide-gap (cf. Table I).



II. HYDRODYNAMIC STABILITY

Introduction

During the early stages of development of the wide-gap machines large scale, irregular convective mixing was observed during experiments with dye tracers. One of the possible causes is the flow reversal caused by buoyancy (see Part I and Ostrach, 1976). At that time, the absence of quantitative data prevented a rest of this hypothesis. Later experiments at General Electric by H. Semon (1977) provided additional data and disclosed a persistent "wavering" of the sample stream at low power inputs; higher power levels caused irregular mixing. However, the power levels corresponded to maximum (centerline) temperatures only a few degrees higher than the wall (buffer) temperature, far below those which could produce the w-shaped profiles described in Part I. According to the analysis in Section I the dimensionless group $g\beta_0 E_0^2 d^4 / \nu_0 k_0 u_0$ must be $O(10^2)$ or more for buoyancy to alter the velocity profile significantly; in the General Electric experiments it was $O(1)$. Changes in the orientation of the flow relative to gravity changed the allowable power levels somewhat but irregular flow persisted at voltage gradients necessary to give a significant electrophoretic separation. It was decided, therefore, to investigate the hydrodynamic stability of the flow.

Several sorts of phenomena are included under the general topic of hydrodynamic stability: the inception of motion in an otherwise quiescent system, the transition from one steady laminar flow to another and transition from laminar to turbulent flow. To establish orders-of-magnitude, theories for the stability of a horizontal layer and of a shear flow were reviewed. As a result it appears that neither of the buoyancy mechanisms involved in these

two situations would be able to destabilize the flow.

Previous work on the stability of stratified fluid layers has centered on quiescent layers (cf. Chandrasekhar, 1961; Ostrach, 1964). This body of literature was combed to locate reports of special significance to the problem at hand which is distinguished by the combined processes of volumetric heat generation and fluid motion. Sparrow, Goldstein and Jonsson (1964) studied the buoyancy driven instability of a quiescent, horizontal layer bounded above and below by rigid walls and heated internally. Although a nonlinear profile does lower the critical temperature difference considerably, calculations based on parameter values for the wide-gap chamber showed that the critical difference is about 10°C which is well above the values reported by General Electric for the horizontal (or vertical) configuration with flow. Allowance for the effect of temperature on the rate of heat generation (which is not part of the Sparrow-Goldstein-Jonsson theory) may lower the critical temperature difference somewhat but an extension of this sort was not attempted, since it seemed best to understand the behavior of the vertical configuration which would be used in electrophoretic separation.

Vest and Arpaci (1969) studied the stability of natural convection in a vertical slot, where however, since the base flow is driven by an anti-symmetric temperature, it is different from that in a continuous flow electrophoresis chamber. The stability with respect to roll waves, oriented perpendicular to the main flow was examined and a critical Grashof number of 7880 was found for this sort of instability. In the General Electric experiment the Grashof number based on the maximum centerline temperature is roughly 10 and, although the circumstances are quite dissimilar, it seems unlikely

that the observed meandering derives from a shear instability of the sort studied by Vest and Arpaci.

If the meandering and subsequent large scale convection are the result of an instability, then it must be one where the critical Rayleigh number is small. One possibility is that the instability derives from small axial temperature gradients which result from uneven heating or cooling. In addition the thermal region near the entrance to the electrode section extends over a region of several chamber half-thicknesses. This gradient can be estimated from an earlier equation describing the balance between convection, conduction and generation,

$$Pe u(y) \frac{\partial \theta(0)}{\partial x} = \frac{\partial^2 \theta(0)}{\partial y^2} + 1 + \sigma_1 \theta(0) \quad (8)$$

for temperature-independent properties. This equation has solutions which decay exponentially with x and have the form

$$e^{-\lambda_n^2 x Pe^{-1}} f_n(y)$$

The mode which has the smallest eigenvalue, λ_n , fixes the relaxation distance, x_e . An estimate of the smallest eigenvalue can be found from the problem where the variable velocity $u(y)$ is approximated by the (constant) average velocity. A more exact calculation would refine this estimate but would not change the order-of-magnitude. It is found that the distance over which the temperature adjusts to the ohmic heating is approximately $(Pe)(d)$. For the wide-gap chamber operation at the conditions listed on Table I the Peclet number is about 50. Thus, since the temperature rise here is over 30 degrees, axial gradients at the inlet (and outlet) are of the order of $1^\circ\text{C}/\text{cm}$.

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Stability of a Fully Developed Flow with an Axial Temperature Gradient

The presence of cold fluid above warmer fluid (in the region above the electrodes with the downflow configuration and in the electrode "outlet" region with upflow) is an unstable configuration. To study the stability of the velocity and temperature fields they were modelled with a fully developed axial flow in a rectangular channel. The basic temperature field consists of an axial gradient of magnitude denoted by A, with lateral variations due to the balance between heat generation and conduction through the walls, i.e.,

$$T = T_w + Ax + \Delta T\theta(y,z) \quad (31)$$

If we scale lengths, velocities, etc., as before we obtain

$$\frac{\partial u}{\partial \tau} = -\frac{\partial p}{\partial x} + N_2(1-\beta\Delta T) - \frac{Ra}{Pr Re} x - \frac{Gr}{Re} \theta + \nabla^2 u \quad (32)$$

$$Pr \frac{\partial \theta}{\partial \tau} + \frac{Ra Re}{Gr} u = \nabla^2 \theta + 1 \quad (33)$$

where

$$Gr = g\beta\Delta Td^3/\nu_0^2 \quad \text{Grashof number}$$

$$Ra = g\beta Ad^4/\nu_0 \alpha_0 \quad \text{Rayleigh number}$$

$$Re = u_0 d/\nu_0 \quad \text{Reynolds number}$$

$$Pr = \nu_0/\alpha_0 \quad \text{Prandtl number}$$

with $u = \theta = 0$ on boundaries at $y = \pm 1$, $z = \pm H$.

The mathematical problem is to identify conditions under which temperature and velocity fields other than the steady-state, symmetric forms exist. In particular we are interested in forms with an exponential time dependence, $e^{\omega t}$. The demarcation between stable and unstable flows is $\omega = 0$. Because of the linear structure of the equations we can show that the imaginary part of ω , $I_m(\omega)$, is zero so that the so-called "exchange of stabilities" principle is satisfied and any disturbance will grow exponentially. Thus, we simply look for conditions where $\omega = 0$. The problem is decomposed into the sum of the steady parts u, θ and perturbations \hat{u} and $\hat{\theta}$.

This problem is very similar to one solved earlier by Ostrach (1955). Here those results are extended to include two-dimensional effects and other disturbance planforms. In the early work the instability was identified through a degeneracy in a base flow, now we see that the disturbances are superimposed on a symmetric base flow.

For the disturbance flow, \hat{u} , and temperature, $\hat{\theta}$, we have

$$\begin{aligned} \nabla_{II}^4 \hat{u} &= Ra \hat{u} \\ \hat{\theta} &= Ra^{-1} \nabla_{II}^2 \hat{u} \end{aligned} \tag{34}$$

where ∇_{II}^2 stands for the two-dimensional Laplace operator. $\hat{u} = \hat{\theta} = 0$ on the boundaries. One set of solutions will, of course, simply be multiples of the symmetric (with respect to the x-z and x-y planes) steady-state solutions. We are interested in anti-symmetric solutions, which represent no change in the volumetric flow rate through the y-z plane. In general, the solutions to these equations can be written

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$$\hat{u} = u_1 + u_2 \quad (35)$$

where

$$\nabla_{II}^2 u_1 = \lambda^2 u_1 \quad (36)$$

$$\nabla_{II}^2 u_2 = -\lambda^2 u_2$$

$$\lambda^4 = Ra$$

The temperature is

$$\theta = Ra^{-1/2} (u_1 - u_2), \quad (37)$$

Solutions are

$$\begin{aligned} u_1 = \sin qz (A_1 \sinh \gamma_1 y + B_1 \cosh \gamma_1 y) \\ + \cos qz (A_2 \sinh \gamma_1 y + B_2 \cosh \gamma_2 y) \\ \gamma_1^2 = q^2 + \lambda^2 \end{aligned} \quad (38)$$

$$\begin{aligned} u_2 = \sin qz (A_3 \sinh \gamma_2 y + B_3 \cosh \gamma_2 y) \\ + \cos qz (A_4 \sinh \gamma_2 y + B_4 \cosh \gamma_2 y) \\ \gamma_2^2 = q^2 - \lambda^2 \end{aligned}$$

To satisfy the boundary conditions on the walls at $z = \pm H$ either

$$(i) \quad \sin qH = 0, A_2 = B_2 = A_4 = B_4 = 0, \quad q = n\pi/H; \quad (39)$$

or

$$(ii) \quad \cos qH = 0, A_1 = B_1 = A_3 = B_3 = 0, \quad q = \frac{2n-1}{2} \frac{\pi}{H}. \quad (40)$$

The first condition, (i), corresponds to a disturbance that is anti-symmetric with respect to the x - y plane with upflow on one side and downflow on the

other. This form preserves the volumetric flow rate. The second condition, (ii), describes a flow with the requisite asymmetry if $B_2 = B_4 = 0$.

Next, with (i), either $B_1 = B_3 = 0$ so that $A_3 = 0$ and $\gamma_1 = \pm i n \pi$ which gives

$$\lambda^4 = n^4 \pi^4 (1 + H^{-2})^2 \quad (41)$$

or $A_1 = A_3 = 0$ so that $B_3 = 0$ and $\gamma_1 = i(2n - 1)\pi/2$ which gives

$$\lambda^4 = \pi^4 [(2n - 1)^2/4 + n^2/H^2]^2 \quad (42)$$

With (ii), $A_4 = 0$ and $\gamma_2 = \pm i n \pi$ so that

$$\lambda^4 = \pi^4 [n^2 + (2n-1)^2/4H^2]^2 \quad (43)$$

The mode with the lowest critical Rayleigh number corresponds to the velocity field

$$u_1 = \sin \frac{\pi z}{H} \cos \frac{\pi v}{2} \quad (44)$$

with the critical Rayleigh number

$$Ra_c = \frac{\pi^4}{16} (1 + H^{-2})^2 \quad (45)$$

For the wide-gap chamber with dimensions noted on Table I, $Ra_c = 6.58$ and with a narrow-gap, $Ra_c = 6.11$. Using data for water at 10°C the critical temperature gradient is $0.5^\circ\text{C}/\text{cm}$ for the wide-gap and $53^\circ\text{C}/\text{cm}$ for the narrow-gap chambers. These results agree *qualitatively* with experimental observation in that they disclose a great deal of sensitivity to axial

temperature gradients for the wide-gap machine. With a narrow-gap it would be rather difficult to achieve the axial gradients large enough to excite the instability.

Although this explanation is consistent with experimental findings for a vertical configuration, questions still remain with regard to tilted or horizontal configurations. These cases have not been analyzed in detail but it is easy to show (mathematically) that buoyancy effects coupled with an axial gradient will destroy the unidirectional character of the flow. At present, however, neither experimental data or quantitative theoretical results are sufficient to assess this matter fully.

III. MODELING THE ELECTROPHORETIC SEPARATION IN CONTINUOUS FLOW
ELECTROPHORESIS

Introduction

We turn now to the task of describing how a sample with a particular mobility distribution will be altered in its passage through a continuous flow electrophoresis chamber. To provide a first approximation we have chosen to base the model on one-dimensional approximations to the various flow and temperature fields and ignore, for the present, two-dimensional effects due to side walls at $z = \pm h$ (except insofar as they cause the electro-osmotic flow to recirculate). Consequently, the temperature and velocity fields and the particle mobility are functions of y alone. This procedure is accurate as long as $d/h \ll 1$ and regions near the side walls are ignored. The effect of the side walls requires much more extensive analysis and remains to be done.

The model for separation is based on the fact that the velocity of a particle of a given mobility can be written as a superposition of an axial velocity, $u(y)$, and a transverse component made up of the electro osmotic flow velocity, $w(y)$ and the particle velocity due to electrophoresis, $v_m(y)$. Thus for a particle of mobility, m , say, the velocity is

$$\hat{i}u(y) + \hat{k}[w(y) + v_m(y)] \quad (46)$$

and particles which enter with the sample at y_0, z_0 will exit at $y = y_0$, $z = z_0 + L[w(y) + v_m(y)]/u(y)$. If we denote the mobility distribution as $N_m(x, y, z)$, to represent the number density of particles with mobility, m , then

$$u(y) \frac{\partial N_m}{\partial x} + [w(y) + v_m(y)] \frac{\partial N_m}{\partial z} = 0 \quad (47)$$

describes the fact that the particles are conserved. The number density at a point x, y, z , is, therefore,

$$N_m\{0, y, z - x[w(y) + v_m(y)]/u(y)\}$$

and the problem is simply one of "tracking".

To predict the mobility distributions at the exit from the separator we suppose that there are a number of collector tubes of area $(2d)(\Delta)$ at the outlet plane. The mobility distribution in a given collector is simply

$$\langle N_m \rangle_i = \frac{1}{\Delta} \int_{z_i}^{z_i + \Delta} \bar{N}_m(z) dz \quad (48)$$

$$\bar{N}_m(z) = \frac{\int_{-d}^d N_m(L, y, z) u(y) dy}{\int_{-d}^d u(y) dy} \quad (49)$$

A computer program was written to implement this scheme; representative results are tabulated in Table II.

Outline of the Computation Method

The computation of the mobility distribution in the collection streams proceeds as follows

Main Input Data are:

Chamber dimensions ($2d$ and $2h$)

Electrode length (ℓ)

Number of collection streams

Electric field strength

Buffer flowrate

Buffer temperature

Constants in the linear equation for buffer electrical conductivity

Constants in the linear equation for buffer thermal conductivity

Buffer conductivities for heat and electricity at the wall temperature

Coating mobility at 20°C

Mobility distribution of sample at 20°C

Location and size of sample stream.

The program then evaluates the temperature field using Eqs. (7), (13) and (15) and calculates the local values of density and viscosity using analytical formulas supplied. (Other relations can be used if necessary.) Then the local axial and electro-osmotic velocities are calculated using the appropriate equations for temperature dependent properties cited in Part I, Eqs. (23) and (30). Finally, particles on the edge of the sample are tracked to the outlet, using mobilities which reflect the local temperature, and the mobility distribution for each collector calculated.

Numerical output includes

Temperature field

Axial velocity field

Electro-osmotic velocity field

Mobility distribution for each collector.

Results of two representative calculations, one with a narrow-gap chamber, the other with a wide-gap, are shown on Table II. General conditions are as shown on Table I. The sample contained two types of particles in equal amounts. Although both configurations show complete separation the wide-gap configuration separates the sample into two widely spaced collectors; the narrow-gap chamber barely separates the two fractions and if either fraction contained a distribution of mobilities there would be overlap. It was not possible to operate the wide-gap chamber model at 1-g without the recirculating flows described in Part II - so gravity had to be suppressed in the calculation.

These calculations are intended to be *illustrative* of the sorts of results that can be obtained using the models derived here. Much more extensive calculations are required to establish the differences in the separatory capabilities of various continuous flow devices.

TABLE I.

Parameters used in numerical calculations

Fluid Properties (A-1 Buffer)

Buffer Temperature	10	°C
Density	1.0	g/cm ³
Viscosity	1.33×10^{-2}	g/cm-s
Thermal Conductivity	5.82×10^{-3}	watts/cm-°C
Electrical Conductivity	6.9×10^{-4}	(ohm-cm) ⁻¹
Coefficient of Expansion	8.62×10^{-5}	(°C) ⁻¹
Thermal Conductivity Coefficient	2.58×10^{-3}	(°C) ⁻¹
Electrical Conductivity Coefficient	3.12×10^{-2}	(°C) ⁻¹

Chamber Parameters

	<u>Narrow-Gap</u>	<u>Wide-Gap</u>
Electric Field Strength, v/cm	70	70
Gravitational Constant*, cm/s ²	980	980
Gap Distance (2d), cm	0.15	0.5
Width (2h), cm	5	5
Length, cm	16	10
Volumetric Flow, cm ³ /s	0.35	0.7
Average Velocity, cm/s	0.467	0.28
ΔT_T , °C	3.17	35.2
Re	2.63	5.26
N ₂	8.88×10^2	1.64×10^4
N ₃	2.4×10^{-1}	4.98×10^1
k ₁	8.2×10^{-3}	9.1×10^{-2}
σ_1	9.9×10^{-2}	1.1

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* except where noted

TABLE II

Red-Blood Cell Separation in Narrow-Gap and Wide-Gap Machines

To demonstrate the use of the separation model two runs were made at conditions shown on Table I with samples made up of equal amounts of particles ('red-blood cells') with mobilities of 2.15 $\mu\text{m-cm/v-s}$ ($m=1$) and 4.15 $\mu\text{m-cm/v-s}$ ($m=2$) (at 20°C).

Other conditions were:

Coating mobility 2.15 $\mu\text{m-cm/v-s}$

Sample inlet size 0.05 cm

Number of collectors 100

Locations of Separated Particles

Narrow-Gap Chamber

Chamber #	<u>1-13</u>	<u>14</u>	<u>15</u>	<u>16</u>	<u>17</u>	<u>18</u>	<u>19</u>	<u>20</u>	<u>21-100</u>
m = 1	0	.177	.590	.233	0	0	0	0	0
m = 2	0	0	0	0	.159	.477	.353	.021	0

Wide-Gap Chamber*

Chamber #	<u>1-20</u>	<u>21</u>	<u>22</u>	<u>23</u>	<u>24-29</u>	<u>30</u>	<u>31</u>	<u>32</u>	<u>33-100</u>
m = 1	0	.043	.546	.411	0	0	0	0	0
m = 2	0	0	0	0	0	.282	.610	.108	0

* g = 0

ACKNOWLEDGMENTS

Dr. R.S. Snyder of MSFC, Dr. G.V.F. Seaman of the University of Oregon and Dr. R.N. Griffin of General Electric have each helped us to understand the many problems involved in the electrophoresis of small particles.

Kurt Hebert of Princeton University did much of the computer programming.

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COMPUTER PROGRAMS

Four programs were written in FORTRAN IV:

ZTEMP - the two-dimensional temperature field, Eq. (12)

VELO - the two-dimensional velocity field, Eq. (19)

TRANST - the transient temperature field, Eq. (17)

TEMP - the separation model.

Listings of these are given on the following pages.

ORIGINAL PAGE IS
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FORTRAN IV G LEVEL 21

MAIN

DATE = 78110

09/10/45

```

C
C
C *****
C *
C * ZTEMP DETERMINES THE TEMPERATURE IN THE ELECTROPHORESIS CELL AS A
C * FUNCTION OF BOTH Y AND Z. THROUGH A CALL TO THE PROCEDURE PERVUE, IT
C * THEN PLOTS A PERSPECTIVE VIEW OF THE TEMPERATURE SURFACE. ZTEMP ACCLP0070
C * AS INPUT PARAMETERS VALUES OF THESE VARIABLES--
C *
C * SIGMA1 - PARAMETER IN THE EXPRESSION FOR ELECTRICAL CONDUCTIVITY.
C * BI - THE BICI NUMBER
C * W - THE ASPECT RATIO OF THE CHANNEL
C * NTERMS - THE NUMBER OF TERMS TO BE TAKEN IN THE SERIES FOR TEMPERA-
C * TURE
C * NYPTS - THE NUMBER OF EQUALLY SPACED CONSTANT-Y LINES FROM (AND
C * INCLUDING) Y=0 TO Y=1 ALONG WHICH TEMPERATURE IS EVALUATED.
C * NZPTS - THE NUMBER OF EQUALLY SPACED CONSTANT-Z LINES FROM (AND
C * INCLUDING) Z=0 TO Z=W ALONG WHICH TEMPERATURE IS EVALUATED.
C *
C *****
C
C
C
C
0001 DIMENSION TEMP (201,101),
      X YY (101),
      X ZZ (201)
0002 READ LABELA,LABELB2
0003 100 FORMAT (1H1)
0004 110 FORMAT ('0', 'SIGMA1 =', 1PE9.2,4X, 'NO. TERMS =', I7, 2X,
      X 'ASPECT RATIO =', 1PE9.2/'0', 'BICI NO. =', 1PE5.2, 2X,
      X 'NO. Y POINTS =', I4, 2X, 'NO. Z POINTS =', I4,
0005 101 FORMAT ('0', 9X, '2', 10X, 'TEMPERATURE')
0006 130 FORMAT (' ')
0007 103 FORMAT (2(2X, 1PE14.6))
0008 150 FORMAT (3E10.3, 3I5)
0009 READ(5, 150) SIGMA1, BI, W, NTERMS, NYPTS, NZPTS
0010 PI=3.1415926
C
C
C
C
0011 DY=1./(NYPTS-1.)
0012 CZ=W/(NZPTS-1.)
C
C THE OUTER LOOP INCREMENTS Y, THE NEXT INNERMOST LOOP INCREMENTS Z, AND TH0420
C INNERMOST TWO LOOPS TAKE THE SERIES FOR TEMPERATURE OUT TO NTERMS TERMS. 0430
C THE PROGRAM STORES THE TEMPERATURE IN THE ARRAY TEMP(I,J), WITH INDICES 10440
C I REFERRING TO THE Y-DIRECTION COORDINATE AND Z-DIRECTION COORDINATE 0450
C RESPECTIVELY. THE ACTUAL Y AND Z COORDINATES ARE STORED IN ARRAYS YY AND 0460
C
0013 Y=-DY
0014 DO 30 I=1, NYPTS
0015 Y=Y+DY
0016 A=PI*(1.+Y)/2.
0017 Z=-CZ
0018 DO 20 J=1, NZPTS
0019 Z=Z+CZ
0020 SUB=C.
0021 DO 10 K=1, NTERMS

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PORTMAN IV G LEVEL 21

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0022          N=2*K-1                                J570
0023          LAMBDA2=N*N*Z*I*PI/4.-SIGMA1          0580
0024          T=(1.-(-1.)**N)/N/LAMBDA2           0590
0025          LAMBDA=LAMBDA2**0.5                   0600
0026          AE1=LAMBDA*(Z-W)                       J610
0027          AE2=-LAMBDA*(Z+W)                     0620
0028          AE3=-2.*LAMBDA*W                      J630
0029          IF (AE1+50.) 1,1,2                     J640
0030          1          E1=0.                        0650
0031          GC TO 3                                  J660
0032          2          E1=EXP(AE1)                   J670
0033          3          IF (AE2+50.) 4,4,5           0680
0034          4          E2=0.                        J690
0035          GC TO 6                                  0700
0036          5          E2=EXP(AE2)                   0710
0037          6          IF (AE3+50.) 7,7,8           0720
0038          7          E3=0.                        0730
0039          GC TO 9                                  0740
0040          8          E3=EXP(AE3)                   0750
0041          9          CONTINUE                      0760
0042          I'=(E1+E2)/(LAMBDA*(1.-E3)-E1*(1.+E3)) 0770
0043          Z=C+T*BI*T1                              J780
0044          10         SU1=SU1+B*SIN(N*K)           0790
0045          TEMP(J,I)=2.*SU1/PI                     J800
0046          20         ZZ(J)=Z                      0810
0047          30         YY(I)=Y                      0820
0048          IEND=I-TIME(1,1)                        0830
C                                                    0840
C PLCI THE TEMPERATURE FIELD AS A SURFACE VIEWED IN PERSPECTIVE. 0850
C                                                    0860
0049          K=2*NYPTS                                0870
0050          DC 60 I=1, NYPTS                          0880
0051          K=K-1                                    0890
0052          L=2*NZPTS                                0900
0053          DC 50 J=1, NZPTS                          0910
0054          L=L-1                                    0920
0055          50         TEMP(L,K)=TEMP(NZPTS+1-J, NYPTS+1-L) 0930
0056          60         CONTINUE                      0940
0057          NYM=NYPTS-1                              0950
0058          NZM=NZPTS-1                              0960
0059          K=2*NYPTS                                0970
0060          DC 62 I=1, NYPTS                          0980
0061          K=K-1                                    0990
0062          DC 62 J=1, NZM                            1000
0063          62         TEMP(NZPTS-J,K)=TEMP(NZPTS+J,K) 1010
0064          NY=L*NYPTS-1                              1020
0065          NZ=L*NZPTS-1                              1030
0066          DC 65 J=1, NZ                             1040
0067          DC 65 I=1, NYM                            1050
0068          65         TEMP(J, NYPTS-I)=TEMP(J, NYPTS+I) 1060
0069          Z=-T2-W                                    1070
0070          DC 70 J=1, NZ                             1080
0071          Z=Z+I2                                    1090
0072          70         ZZ(J)=Z                      1100
0073          Y=-LY-1.                                  1110
0074          DC 80 I=1, NY                             1120

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0075          Y=Y+CY                      1130
0076      8)  YY(1)=Y                      1140
0077          JT=IFNE1*10.+1.             1150
0078          NCONTR=JT*2                  1160
0079          TMAX=JT/10.                  1170
C
C      PRINT THE FIELD FOR Y=0.           1180
C
0080          WRITE (6,100)                 1210
0081          WRITE (6,110) SIGMA1,NTERMS,N,BI,NYPTS,NZPTS 1220
0082          WRITE (6,101)                 1230
0083          WRITE (6,102)                 1240
0084          DO 40 J=1,NZ                  1250
0085          40  WRITE (6,103) ZZ(J),TEMP(J,NYPTS) 1260
C
C      PRINT THE FIELD FOR Z=0.           1270
C
0086          WRITE (6,102)                 1290
0087      102 FORMAT ('J',9X,'Y',10X,'TEMPERATURE') 1300
0088          DO 41 I=1,NY                  1310
0089      41  WRITE (6,103) YY(I),TEMP(NZPTS,I) 1320
0090          CALL PELVUF (11,ZZ,YY,1E*E,4*0.5,4.,TMAX+1.5,201,101,NZ,NY,C, 1340
X NCONTR,J,-.1,TMAX,0,0,0,9.,4.,-1,17,'TEMPERATURE FIELD') 1350
0091          STOP                           1360
0092          END                             1370

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FORTRAN IV G LEVEL 21

MAIN

DATE = 78110

03/14/24

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C
C
C
C *****
C *
C * VELO DETERMINES THE DOWNWARD VELOCITY IN THE ELECTROPHORESIS CELL AS A
C * FUNCTION OF BOTH Y AND Z. THROUGH A CALL TO THE PROCEDURE PERVUE, IT
C * THEN PLOTS A PERSPECTIVE VIEW OF THE VELOCITY SURFACE.
C *
C *   SIGMA1  -PARAMETER IN THE EXPRESSION FOR THE ELECTRICAL
C *             CONDUCTIVITY
C *   SI      -THE SIOT NUMBER
C *   W       -THE ASPECT RATIO OF THE CHANNEL
C *   NTERMS  -THE NUMBER OF TERMS TO BE TAKEN IN THE SERIES FOR THE
C *             VELOCITY
C *   NYPTS   -THE NUMBER OF EQUALLY SPACED CONSTANT-Y LINES FROM (AND
C *             INCLUDING) Y=0 TO Y=1 ALONG WHICH THE VELOCITY IS
C *             EVALUATED
C *   NZPTS   -THE NUMBER OF EQUALLY SPACED CONSTANT-Z LINES FROM (AND
C *             INCLUDING) Z=0 TO Z=W ALONG WHICH VELOCITY IS EVALUATED
C *
C *****
C
C
C
C
0001   DIMENSION U(201,101),YY(101),ZZ(201)
0002   REAL BASEDA,LAMBDA2,N2,N3
0003   100  FORMAT (1H1)
0004   110  FORMAT ('1', 2X, 'SIGMA1 =',E10.3,2X, 'BIOT NO. =',E10.3,2X,
0005   X 'ASPECT RATIO =',E10.3,2X, 'N2 =',E10.3,2X, 'N3 =',E10.3)
0006   111  FORMAT ('1', 2X, 'NTERMS =',I5,2X, 'NYPTS =',I5,2X, 'NZPTS =',I5)
0007   120  FORMAT ('1', 9X, 'Z', 11X, 'X VELOCITY')
0008   130  FORMAT (' ')
0009   140  FORMAT (2(2X,1P=14.0))
0010   150  FORMAT (5E10.3,3I5)
0011   160  FORMAT ('0', 9X, 'Y', 11X, 'X VELOCITY')
0012   READ (5,15),END=999) SIGMA1,SI,=.N2,N3,NTERMS,NYPTS,NZPTS
0013   WRITE (6,100)
0014   WRITE (6,111) SIGMA1,SI,W,N2,N3
0015   WRITE (6,111) NTERMS,NYPTS,NZPTS
0016   DMIN=0.
0017   DMAX=0.
0018   F1=3.1415926
0019   F12=F1*PI
0020   F124=F12/4.
0021   F14=F1/4.
0022   F1W=PI*W
0023   WF12=F1*W/2.
0024   TPI=2./PI
0025   SUM1=0.
0026   SUM2=0.
0027   SUM3=0.
0028   K=-1
C
C DETERMINE THE CONSTANT, K, FROM THE INTEGRATED VELOCITY.
C
0028   DO 5 N=1,NTERMS

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PORTBAN IV G LEVEL 21

MAIN

DATE = 76110

09/14/24

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0029      K=K+2                                0570
0030      LAMBDA2=K*K*PI24-SIGMA1              0580
0031      LAMBDA=LAMBDA2**0.5                  0590
0032      FN2=K*K                               0600
0033      FN4=FN2*FN2                           0610
0034      F1=1/F2/K                             0620
0035      PTANH1=TANH(Y*FPI2)                   0630
0036      PTANH2=TANH(LAMBDA*W)                 0640
0037      S1=(21*PTANH1-W)/FN4                  0650
0038      S2=S1/LAMBDA2                         0660
0039      S3=((P1*PTANH1-PTANH2/LAMBDA)/(LAMBDA*PTANH2-BI))/FN2/LAMBDA2 0670
0040      SUM1=SUM1+S1                           0680
0041      SUM2=SUM2+S2                           0690
0042      SUM3=SUM3+S3                           0700
0043      S CONTINUE                             0710
0044      A1=SUM1*32./W/(PI**4)                 0720
0045      A2=SUM2*32./W/(PI**4)                 0730
0046      A3=SUM3*32./SIGMA1/FI/PI/W           0740
0047      FK=(1.+A2*A1-N3*A2-N3*BI*A3)/A1       0750
0048      WPI2 (6,112) FK                       0760
0049      112 FORMAT (' ',2X,E10.3)            0770
C
C EVALUATE VELOCITY AT EACH (Y,Z) POINT TAKEN.
C
0050      DY=1./(NYPTS-1.)                      0810
0051      DZ=W/(NZEIS-1.)                      0820
0052      Y=-DY                                  0830
0053      FK1=FK-N2                              0840
0054      F1=8./PI2                              0850
0055      F2=2.*N3*EI/SIGMA1                    0860
0056      DO 30 I=1,NYPTS                        0870
0057      Y=Y+DY                                  0880
0058      Z=F1*(1.+Y)/2.                         0890
0059      Z=-DZ                                  0900
0060      DO 20 J=1,NZPTS                         0910
0061      Z=L+DZ                                  0920
0062      SUM=0.                                  0930
0063      K=-1                                    0940
0064      DO 10 N=1,NIERMS                       0950
0065      K=K+2                                    0960
0066      LAMBDA2=K*K*PI24-SIGMA1              0970
0067      LAMBDA=LAMBDA2**0.5                  0980
0068      FN3=K**3                              0990
0069      PTANH=TANH(LAMBDA*W)                 1000
0070      C1=F1*(FK1+N3/LAMBDA2)/FK3          1010
0071      C3=F2/K/LAMBDA2                      1020
0072      C2=C3/(LAMBDA*PTANH-BI)             1030
0073      ARG1=K*(Z-W)/TPI                     1040
0074      ARG2=-K*(Z+W)/TPI                   1050
0075      ARG3=-K*FI                           1060
0076      ARG4=LAMBDA*(Z-W)                   1070
0077      ARG5=-LAMBDA*(Z+W)                  1080
0078      ARG6=-2.*LAMBDA*W                   1090
0079      IF (ARG1.LE.-100.) E1=0.             1100
0080      IF (ARG1.GT.-100.) E1=EXP(ARG1)     1110
0081      IF (ARG2.LE.-100.) E2=0.             1120

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FORTRAN IV G LEVEL 21

MAIN

DATE = 7/11/70

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0082      IF (ARG2.GT.-100.) E2=EXP(ARG2)      1130
0083      IF (ARG3.LE.-100.) E3=.             1140
0084      IF (ARG3.GT.-100.) E3=EXP(ARG3)      1150
0085      IF (ARG4.LE.-100.) E4=.             1160
0086      IF (ARG4.GT.-100.) E4=EXP(ARG4)      1170
0087      IF (ARG5.LE.-100.) E5=.             1180
0088      IF (ARG5.GT.-100.) E5=EXP(ARG5)      1190
0089      IF (ARG6.LE.-100.) E6=.             1200
0090      IF (ARG6.GT.-100.) E6=EXP(ARG6)      1210
0091      E=(E1+E2)/(1.+E3)                    1220
0092      A1=C1+E                               1230
0093      A2=C2+E                               1240
0094      A3=-C3*(E4+E5)/(LAMBDA*(1.-E6)-E1*(1.+E6)) 1250
0095      A4=-C1                               1260
0096      SUM=SUM+(A1+A2+A3+A4)*SIN(A*X)        1270
0097      CONTINUE                             1280
0098      U(J,I)=-2.*SUM/PI                    1290
0099      IF (J(J,I).LE.UMIN) UMIN=U(J,I)      1300
0100      IF (U(J,I).GE.UMAX) UMAX=U(J,I)      1310
0101      Z(J)=Z                               1320
0102      Y(I)=Y                               1330
C
C      PLOT THE VELOCITY FIELD AS A SURFACE VIEWED IN PERSPECTIVE.
C
0103      K=2*NYPTS                              1340
0104      DO 60 I=1, NYPTS                      1350
0105      A=K-1                                  1360
0106      L=2*NZPTS                              1370
0107      DO 50 J=1, NYPTS                      1380
0108      L=L-1                                  1390
0109      U(L,K)=U(NYPTS+1-J, NYPTS+1-I)      1400
0110      CONTINUE                             1410
0111      NY=NYPPTS-1                          1420
0112      NZ=NZPTS-1                          1430
0113      K=2*NYPTS                              1440
0114      DO 62 I=1, NYPTS                    1450
0115      K=K-1                                  1460
0116      DO 62 J=1, NZ                        1470
0117      U(NZPTS-J, K)=U(NZPTS+J, K)          1480
0118      NZ=2*NZPTS-1                          1490
0119      NY=2*NYPTS-1                          1500
0120      DO 65 J=1, NZ                        1510
0121      DO 65 I=1, NY                       1520
0122      U(J, NYPTS-I)=U(J, NYPTS+I)        1530
0123      U(1,1)=0.                             1540
0124      U(NZ,1)=0.                            1550
0125      U(1,NY)=0.                           1560
0126      U(NZ,NY)=0.                          1570
0127      Z=-DZ-W                               1580
0128      DO 70 J=1, NZ                        1590
0129      Z=Z+DZ                               1600
0130      Z(J)=Z                               1610
0131      Y=-DY-1.                             1620
0132      DO 80 I=1, NY                       1630
0133      Y=Y+DY                               1640
0134      Y(I)=Y                               1650

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FORTRAN IV C LEVEL 21

HAIN

DATE = 76110

09/14/24

0135	WRITE (6,120)	1690
0136	WRITE (6,130)	1700
0137	DO 40 I=1,NZ	1710
0138	40 WRITE (6,140) ZZ(I),U(I,NYPTS)	1720
0139	WRITE (6,160)	1730
0140	WRITE (6,130)	1740
0141	DO 90 I=1,NY	1750
0142	90 WRITE (6,140) YY(I),U(NZPTS,I)	1760
0143	UMIN=1.1*UMIN	1770
0144	UMAX=1.1*UMAX	1780
0145	CALL PERVE(11,ZZ,YY,U,U+.5,3.,3.,201,101,NZ,NY,0,10,0,UMIN,UMAX, X 0,0,0,9.,9.,-1,14,'VELOCITY FIELD')	1790
0146	999 STOP	1810
0147	END	1820

FORTRAN IV G LEVEL 21

MAIN

DATE = 7E104

15/00/50

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C
C
C .....
C *
C * TRANSPI DETERMINES THE AVERAGE TEMPERATURE AND CENTER TEMPERATURE OF
C * THE ELECTROPHORESIS CHANNEL AS FUNCTIONS OF TIME. THE REQUIRED INPUT
C * DATA IS LISTED IN THE OUTPUT OF THE PROGRAM. ALL THE INPUT VARIABLES
C * SELF-EXPLANATORY EXCEPT THE FOLLOWING-
C *
C * NUNTAU - THE NUMBER OF TIMES AT WHICH TEMPERATURE IS EVALUATED.
C * CTAU - THE DIMENSIONLESS TIME TO WHICH TEMPERATURE IS EVALUATED
C * NTERMS - THE NUMBER OF TERMS TAKEN FOR TAU = 0 IN THE SERIES FOR
C * TEMPERATURE. FOR TAU GREATER THAN ZERO, THE NUMBER OF TERMS
C * TAKEN IS CONTROLLED BY NUMBER.
C .....
C
C
0001 DIMENSION CTMP(1001)
0002 DIMENSION AVTEMP(1001)
0003 DIMENSION T(1001)
0004 DIMENSION TIME(1001)
0005 DIMENSION NUMBER(1001)
0006 101 FORMAT (7E10.3)
0007 102 FORMAT (215,2E10.3)
0008 103 FORMAT (1H1)
0009 104 FORMAT ('C', 'THERMAL CONDUCTIVITY -', 1PE10.2, 1X, 'WATTS/CM-C', 2X,
X 'DENSITY -', 1PE10.2, 1X, 'G/CC', 2X, 'SPECIFIC HEAT -', 1PE10.2
X 'CAL/C-G'/'C', 'ELECTRIC CONDUCTIVITY -', 1PE10.2, 1X,
X '/OHM-CM', 2X, 'FIELD STRENGTH -', 1PE10.2, 1X, 'V/CM'/'C',
X 'WELL DEPTH -', 1PE10.3, 1X, 'CM', 2X, 'WELL TEMP. -', 1PE10.3
X 'C')
0010 105 FORMAT ('C', 'NUNTAU =', 17, 2X, 'NTERMS =', 17, 2X, 'SIGMA1 =', 1PE10.2,
X 'CTAU =', 1PE10.3)
0011 106 FORMAT ('C', 6X, 'ELCOND', 9X, 'AVTEMP-C', 6X, 'CTEMP-C')
0012 107 FORMAT (' ')
0013 108 FORMAT (3(2X, 1PE14.6))
0014 READ (5,101) THCOND, DENSITY, SPHT, ELCOND, IO, DEPTH, TBUFF
0015 READ (5,102) NUNTAU, NTERMS, SIGMA1, CTAU
0016 WRITE (6,103)
0017 WRITE (6,104) THCOND, DENSITY, SPHT, ELCOND, IO, DEPTH, TBUFF
0018 WRITE (6,105) NUNTAU, NTERMS, SIGMA1, CTAU
0019 FN1=SIGMA1**0.5
0020 CSN=CLS(FN1)
0021 PI=3.1415926
0022 A=(1./CSN-1.)/SIGMA1
0023 B=TAN(FN1)/FN1/SIGMA1-1./SIGMA1
0024 CTAU=CTAU/(NUNTAU-1)
C
C COMPUTE THE TIME AND TEMPERATURE SCALES FROM THE DIMENSIONED INPUT VARIABLES
C
0025 TIME=ELCOND*IO*DEPTH*DEPTH/THCOND
0026 TIME=4*DEPTH*DEPTH*DENSITY*SPHT/THCOND*4.104
C
C NUMBER SPECIFIES THE NUMBER OF TERMS TAKEN IN THE SERIES FOR TEMPERATURE,

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FORTRAN IV G LEVEL 21

MAIN

DATE = 76104

15/00/50

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C FOR TAU GREATER THAN ZERO. IF MORE THAN NUMBER TEMPS ARE TAKEN, LAPCALS(107)
C UNDERFLOW MAY OCCUR.
C
0027      HC=NUMTAU-1
0028      DO 4 J=1,HC
0029          Z=J+1
0030          TAU=(I-1)*DTAU
0031      +   NUMBER(1)=.5*((16./TAU)**.5-1.)
0032      NUMBER(1)=NTEMP5
C
0033      TAU=-DTAU
0034      DO 7 JTAU=1,NUMTAU
0035          TAU=TAU+DTAU
0036          AVSUM=0.
0037          CSUM=C.
0038          N1=1+NUMBER(JTAU)
0039          DO 6 N=1,N1
0040              M=N-1
0041              TN=2*M+1
0042              TN2=TN*TN
0043              C1=EXP((-TN2*PI*PI+4.*SIGNA1)*TAU)
0044              L=(4.*SIGNA1-TN2*PI*PI)
0045              AVSUM=AVSUM+C1/TN2/C
0046              CSUM=CSUM+EX*(-1.)**N/E/TN
0047          AVTEMP(JTAU)=S+J2./EI/PI*AVSUM
0048          AVTEMP(JTAU)=TEMPF+TEMPSCI*AVTEMP(JTAU)
0049          CTEMP(JTAU)=A+16./PI*CSUM
0050          CTEMP(JTAU)=TEMPF+TEMPSCI*CTEMP(JTAU)
C
C PRINT AND PLOT (THROUGH DEIPS) THE TEMPERATURE AS A FUNCTION OF TIME.
C
0051      WRITE(6,106)
0052      WRITE(6,107)
0053      TAU=-DTAU
0054      DO 5 K=1,NUMTAU
0055          TAU=TAU+DTAU
0056          TA=TAU*TIMSCI
0057          TIME(K)=TA
0058      5   WRITE(6,108) TIME(K),AVTEMP(K),CTEMP(K)
0059      SX=7.5
0060      CALL DEIPS1(TIME,CTEMP,NUMTAU,01,SX,
X          ' (RELAXATION) ',),(' SECONDS' ),0,
X          ' (LINES CENTIGRADES) ',0)
0061      CALL DEIPS2(TIME,AVTEMP,NUMTAU,1)
0062      CALL DEIPS3(3.,5.,2,(' AVERAGE' ),0)
0063      CALL DEIPS3(3.,7.5,2,(' CENTER' ),0)
0064      STOP
0065      END

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C ..... 0010
C * TEMP EVALUATES AS FUNCTIONS OF Y THE TEMPERATURE PROFILE AND X- AND 0020
C * Z- DIRECTION VELOCITY PROFILES FOR AN ELECTROPHORESIS CELL WITH HEAT 0030
C * CONDUCTION IN THE Y-DIRECTION. IT ALSO MODELS PARTICLE SEPARATION FOR 0040
C * AN INPUT STREAM OF GIVEN SIZE, LOCATION AND MOBILITY DISTRIBUTION. 0050
C * THERE ARE TWO SUBROUTINES- SIMPS, WHICH PERFORMS INTEGRATION BY 0060
C * SIMPSON'S RULE, AND TEMP, WHICH COMPUTES THE TEMPERATURE FIELD. 0070
C * AMONG THE INPUT VARIABLES ARE-- 0080
C * 0090
C * 0100
C * 0110
C * NVEL -NUMBER OF POINTS AT WHICH THE VELOCITY IS EVALUATED 0120
C * NVEL MUST BE ODD 0130
C * NCOL -THE NUMBER OF COLLECTION UNITS 0140
C * NPCOL -THE NUMBER OF POINTS IN EACH COLLECTOR 0150
C * NPCOL MUST BE ODD 0160
C * NM -THE NUMBER OF POINTS TAKEN IN THE MOBILITY HIST. 0170
C * MOBIL -ARRAY REPRESENTING MOBILITIES 0180
C * NSUM -NUMBER OF TERMS TAKEN IN THE INFINITE SERIES FOR THE 0190
C * FIRST ORDER TEMPERATURE FIELD 0200
C * FMOBL -ARRAY REPRESENTING THE MOBILITIES 0210
C * FMOCBL -ARRAY REPRESENTING THE MOBILITY CONCENTRATIONS. 0220
C * ..... 0230
C 0240
C 0250
C 0260
0001 DIMENSION T(2001),HU(2001),FMO(1001),FMOC(1001),S1(1001),S1(501) 0270
      A W2(501),B(501),C(501),YI(501),Z1(501),Z2(501),
      X AX(1001),ZP(1001),ZF2(1001),AR(1001),B3(1001), 0280
      A FNEAR(1001), 0290
      X FMOBL( 50),FMOCBL( 50),DMOL(1001) 0300
0002 REAL ZCC M01 0310
C 0320
0003 101 FORMAT (5I5) 0330
0004 102 FORMAT (4E10.3) 0340
0005 103 FORMAT (2E10.3) 0350
0006 104 FORMAT (3E10.3) 0360
0007 105 FORMAT (4E10.3) 0370
0008 106 FORMAT (2E10.3) 0380
0009 110 FORMAT ('I', 'INTEGER INPUT VARIABLES') 0390
0010 111 FORMAT ('I', 'NVEL =', I5, 2X, 'NCOL =', I5, 2X, 'NPCOL =', I5, 2X, 0400
      X 'NM =', I5, 2X, 'NSUM =', I5) 0410
0011 112 FORMAT ('O', ' / 'O', 'BUFFER PROPERTIES (CONSTANT)') 0420
0012 113 FORMAT ('I', 'VISCOSITY =', 1PE10.2, 1X, 'GM/CM-S', 2X, 'DENSITY =', 0430
      X 1PE10.2, 1X, 'G/CC', 2X, 'THERMAL CONDUCTIVITY =', 1PE10.2, 0440
      X 1X, 'WATTS/CM-C' / ' ELECTRIC CONDUCTIVITY =', 1PE10.2, 0450
      X 1X, ' / CM-S' ) 0460
0013 114 FORMAT ('O', ' / 'O', 0470
      X 'BUFFER PROPERTIES (TEMPERATURE VARYING, DIMENSIONLESS)') 0480
0014 115 FORMAT ('O', 'THERM. COND.', 6X, 'ELECT. COND.') 0490
0015 116 FORMAT ('I', 'PKP1 =', 1PE10.3, 2X, 'FS1P =', 1PE10.3) 0500
0016 117 FORMAT ('I', 'PKP2 =', 1PE10.3, 2X, 'FS2P =', 1PE10.3) 0510
0017 118 FORMAT ('O', ' / 'O', 'CHAMBER DIMENSIONS') 0520
0018 119 FORMAT ('I', 'WILTH =', 1PE10.2, 1X, 'CM', 2X, 'LENGTH =', 1PE10.2, 1X, 0530
      X 'CM', 2X, 'DEPTH =', 1PE10.2, 1X, 'CM') 0540
0019 120 FORMAT ('I', 2X, 'MOBILITY (MICRONS-CM/VOLT-S) ', 2X, 'COLLECTOR', 2X, 0550
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X 'MOBILITY DENSITY')
0020 121 FORMAT (' ',1E10.2,20X,1E10.2) 0570
0021 122 FORMAT (' ',1E10.2,20X,1E10.2) 0580
X 'ELECTRIC FIELD STRENGTH -',E10.2,1X,'V/CM',2X 0590
X 'BUFFER TEMPERATURE -',E10.2,1X,'C',2X,'BUFFER FLOW RATE - 0600
X 'E10.2,1X,'CC/S',/' ', 'COATING MOBILITY -',E10.2,1X, 0610
X 'MICRONS-CM/V-S') 0620
0022 123 FORMAT (' ',2X,'MOBILITY (MICRONS-CM/VOLT-S)',2X,'MOBILITY DENSITY 0630
X') 0640
0023 124 FORMAT (' ',2X,'X (CM)',1X,'TEMPERATURE (C)',6X,'U (-S/S)',6X, 0650
X 'W (CM/S)') 0660
0024 125 FORMAT (' ') 0670
0025 126 FORMAT (' ',F5.3,3(3X,1E14.6)) 0680
0026 127 FORMAT (' ',9X,E14.4,11X,15,3X,E14.4) 0690
0027 128 FORMAT (' ', 'PARTICLES OF MOBILITY',1E10.2,2X, 0700
X 'LCST FROM INITIAL POSITION CH I =',E14.5) 0710
0028 129 FORMAT (' ', 'Y1 =',E14.5,2X, 'Z1 =',E14.5,2X, 'R =',E14.5) 0720
0029 130 FORMAT (15) 0730
0030 131 FORMAT (1E10.3) 0740
0031 132 FORMAT (' ', 'GRAVITATIONAL FIELD STRENGTH =',E14.5,'C*/S**2') 0750
0032 133 FORMAT (' ', 'SAMPLE INLET LOCATION AND SIZE'/', 'COORDINATES X 0760
X '-',14,2X,E10.3,2X,'Z-',14,2X,E10.3,2X,'SAMPLE RADIUS ' 0770
X '2X,E10.3) 0780
0033 134 FORMAT (' ',2X,'X',6X,'TEMPERATURE',10X,'U',15X,'W') 0790
0034 135 FORMAT (' ') 0800
0035 READ (5,101) NVL1,NCCL,MPCL1,M3,MSUB 0810
0036 READ (5,102) FSE1,FSP2,FRF1,FK22 0820
0037 READ (5,103) ELCOND,THCOND 0830
0038 READ (5,104) TIBF,DEPTB,XLNG 0840
0039 READ (5,104) Y1,Z1,R 0850
0040 READ (5,105) E0,TEUFF,TCOBL,UBJFF 0860
0041 READ (5,111) GRAV 0870
0042 READ (5,106) (FICBL(I),FNICBL(I),I=1,N) 0880
C
C EVALUATE TEMPERATURE FIELD AS ARRAY T- GIVES TEMPERATURE AT EACH POINTS 0890
C FROM Y=0 TO Y=1. 0900
C 0910
0043 NDF=2*LEVEL-1 0920
0044 NTEMP=2*NDF-1 0930
0045 TSCALE=ELCOND*E0*E0*DEPTB*DEPTH/THCOND/4. 0940
0046 T1=TEUFF 0950
0047 T2=T1**2 0960
0048 T3=T1**3 0970
0049 DENSITY=0.998203/(1.-(0.0042E-05)*T1+(3.5055E-06)*T2-(0.79E-08)*T3) 0980
0050 CALL TEMP(NTEMP,NSUP,FSE1,FRF1,T) 0990
C 1000
C EVALUATE DENSITY AT EACH TEMPERATURE POINT- ARRAY RHO 1010
C 1020
0051 CRHO=0.998203/DENSITY 1030
0052 DO 2 I=1,NTEMP 1040
0053 *T=TEUFF+TSCALE*I 1050
0054 T2=T**2 1060
0055 T3=T**3 1070
0056 RHO(I)=CRHO/(1.-(0.0042E-05)*T+(3.5055E-06)*T2-(0.79E-08)*T3) 1080
0057 2 CONTINUE 1090

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C
C  EVALUATE VISCOSITY AT EVERY OTHER TEMPERATURE POINT- ARRAY MUJ
C
0058      J=-1
0059      ARG =TEUFF-20.
0060      ARG=3.00067/(1.+C.JC82*ARG)
0061      ARG=-7.6039+ARG
0062      MUC=EXP(ARG)
0063      MU1=J.JJ8937
0064      CTMUEL=CTMCBL*MU1/MU0
0065      VISC=MUJ
0066      EC 3 I=1,NDF
0067          J=J+2
0068      ARG=TSSCALE*T(J)+TSUFF-20.
0069      ARG=3.00067/(1.+C.JC82*ARG)
0070      ARG=-7.6039+ARG
0071      FMU(I)=EXP(ARG)/MUJ
0072      3 CONTINUE
0073      WRITE (6,11)
0074      WRITE (6,111)  NVEL,NCCL,MPCC,MI,MSU1
0075      WRITE (6,112)
0076      WRITE (6,113)  VISC,ELNSTY,THCOND,ZICOND
0077      WRITE (6,114)
0078      WRITE (6,115)
0079      WRITE (6,116)  FAP1,FSP1
0080      WRITE (6,117)  FKE2,FSE2
0081      WRITE (6,118)
0082      WRITE (6,119)  WADIR,XLONG,DEPTH
0083      WRITE (6,122)  ID,TLUFF,TRUFF,CTMCBL
0084      WRITE (6,129)  Y1,Z1,R
0085      WRITE (6,132)  GRAV
0086      WRITE (6,123)
0087      WRITE (6,121)  (FMU(I),FMUEL(I),I=1,NDF)
C
C
C  COMPUTE DENSITY INTEGRAL AS ARRAY FMBCI REPRESENTING DENSITY INTEGRALS AT
C  Y=0 TO EVERY OTHER DENSITY-TEMPERATURE POINT
C
0088      DY=1./(NTEMP-1.)
0089      FMBCI(1)=C.
0090      M1=1
0091      M2=3
0092      EC 4 I=2,NDF
0093          A=DY*(MHC(M1)+4.*MHC(M1+1)+MHC(M2))/J.
0094          M1=M2
0095          M2=M2+2
0096          FMBCI(I)=A+FMBCI(I-1)
0097      4 CONTINUE
C
C  COMPUTE Y/FMU INTEGRAL AS ARRAY REPRESENTING Y/FMU INTEGRALS AT Y=0 TO
C  EVERY OTHER VISCOSITY POINT (I.E., EVERY FOURTH DENSITY-TEMPERATURE POINT)
C  FIRST EVALUATE Y/FMU AS ARRAY S1.
C
0098      DY=1./(NDF-1.)
0099      S1(1)=C.
0100      EC 5 I=2,NDF

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0101          Y=DY*(I-1)                                1650
0102          S1(I)=Y/PNU(I)                            1700
0103          5 CCNTINUE                                 1710
0104          U1(I)=C.                                   1720
0105          N1=1                                       1730
0106          N2=3                                       1740
0107          DC 6 I=2,NVEL                               1750
0108          A=DY*(S1(N1)+4.*S1(N1+1)+S1(N2))/3.       1760
0109          N1=N2                                       177
0110          N2=N1+2                                    1780
0111          U1(I)=A+U1(I-1)                            1790
0112          6 CCNTINUE                                 1800
C
C EVALUATE FRHCI/PNU AS ARRAY U2 REPRESENTING FRHCI/PNU INTERPOLATED FROM Y=0
C EVERY OTHER VISCOSITY POINT. FIRST EVALUATE FRHCI/PNU AS ARRAY S1.
C
0113          S1(1)=C.                                    1810
0114          DC 7 I=2,NDF                                1820
0115          S1(I)=FRHCI(I)/PNU(I)                      1830
0116          7 CCNTINUE                                 1840
0117          J2(1)=C.                                   1850
0118          N1=1                                       1860
0119          N2=N1+2                                    1870
0120          DC 8 I=2,NVEL                               1880
0121          A=DY*(S1(N1)+4.*S1(N1+1)+S1(N2))/3.       1890
0122          N1=N2                                       1900
0123          N2=N1+2                                    1910
0124          J2(I)=A+U2(I-1)                            1920
0125          8 CCNTINUE                                 1930
C
C CHANGE J1 AND U2 FROM INTEGRALS TAKEN FROM ZERO TO Y TO THEIR POINT IN THE
C VELOCITY EXPRESSION- TAKEN FROM Y TO 1.
C
0126          DC 11 I=1,NVEL                              1940
0127          U1(I)=U1(NVEL)-U1(I)                       1950
0128          11 U2(I)=U2(NVEL)-U2(I)                    1960
C
C EVALUATE THE CONSTANT FK BY INTEGRATING -FK*U1 + S1*U2 FROM Y=0 TO Y=1
C AND SETTING RESULT EQUAL TO ONE.
C
0129          DY=1./(NVEL-1.)                             1970
0130          N1=1                                         1980
0131          CALL SIMPS(DY,U1,N1,NVEL,A1)                 1990
0132          CALL SIMPS(DY,U2,N1,NVEL,A2)                 2000
0133          PNU=VISC/DENSTY                             2010
0134          USCAL1=QBUFF/WIDTH/DEPTH                    2020
0135          FN2=GRAV*DEPTH*DEPTH/PNU/USCAL1/4.0         2030
0136          FK=(-1.+FN2*FN2)/A1                          2040
C
C EVALUATE THE TOTAL VELOCITY
C
0137          USUM=).                                      2050
0138          DC 9 I=1,NVEL                                2060
0139          J1(I)=-FK*U1(I)+FN2*U2(I)                   2070
0140          9 USUM=USUM+U(I)                             2080
C

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C COMPUTE THE ELECTROSTATIC (Z-DIRECTION) VELOCITY. THIS HAS THE FORM OF
C SO MERELY USE INTEGRATED U1 TO DETERMINE CONSTANT, AND THEN SUBTRACT 1.
C
0141      USCALE=CTNCBI*EC*.0001
0142      PK2=-1./A1
0143      NSUB=0.
0144      J=-1
0145      DO 12 I=1,NVEL
0146      J=J+2
0147      CMCEI(I)=0.0001*RU1/(RUC*PNU(J))
0148      W(I)=-PK2*U1(I)-1.
0149      12      NSUM=NSUB+W(I)
C
C PRINT TEMPERATURE AND U- AND W- VELOCITY FIELDS AT VALUES OF Y CORRESPONDING
C TO THE VELOCITY POINTS.
C
0150      DY=1./(NVEL-1)
0151      WRITE (6,135)
0152      WRITE (6,134)
0153      WRITE (6,125)
0154      DO 30 I=1,NVEL
0155      Y=DY*(I-1)
0156      J=1+(I-1)*4
0157      30 WRITE (6,126) Y,T(J),U(I),W(I)
0158      DY=DY*DEFTL/2.
0159      WRITE (6,135)
0160      WRITE (6,124)
0161      WRITE (6,125)
0162      DO 10 I=1,NVEL
0163      Y=DY*(I-1)
0164      J=1+(I-1)*4
0165      U(I)=U(I)*USCALE
0166      W(I)=W(I)*USCALE
0167      T(J)=T(J)*ISCALE+TEUFF
0168      10      WRITE (6,126) Y,T(J),U(I),W(I)
C
C
0169      J=2*NVEL
0170      DO 14 I=1,NVEL
0171      J=J-1
0172      CMOEL(J)=CMOEL(NVEL+1-I)
0173      U(J)=U(NVEL+1-I)
0174      14      W(J)=W(NVEL+1-I)
0175      NVELN=NVEL-1
0176      DO 15 I=1,NVELN
0177      CMCEL(NVEL-I)=CMOEL(NVEL+I)
0178      U(NVEL-I)=U(NVEL+I)
0179      15      W(NVEL-I)=W(NVEL+I)
0180      NV=2*NVEL-1
0181      YE=LEFT/2.
0182      Y=-LE-YY
0183      DO 16 I=1,NV
0184      Y=Y+DY
0185      YY(I)=Y
0186      16 CONTINUE

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C   SET THE Z-COORDINATE ARRAY AND THE Z-DIRECTION STEP SIZE.
C
0187      ZCCL=WIDTH/NCOL
0188      DZ=ZCCL/(NFCCL-1)
0189      NZ=(NFCCL-1)*NCCL+1
0190      Z=WIDTH/2.
0191      NYPTS=NVEL
0192      NZPTS=(NZ-1)/2+1
0193      Z=-ZD-DZ
0194      DO 190 I=1,NZ
0195      Z=Z+DZ
0196      190 ZZ(I)=Z
C
C   ROUND THE CENTER AND RADIUS OF THE SAMPLE STREAM TO THE NEAREST MULTIPLE
C   OF DY.
C
0197      IY1=(Y1/DY+.5)*NYPTS
0198      IZ1=(Z1/DZ+.5)*NZPTS
0199      Y1=YY(.Y1)
0200      Z1=ZZ(.IZ1)
0201      XR=R/DY+.5
0202      R=NR*DY
0203      WRITE (6,133) IY1,Y1,IZ1,Z1,R
C
C   FOR EACH MOBILITY TAKEN IN THE SAMPLE DISTRIBUTION, DETERMINE THE EXIT
C   LOCATIONS OF PARTICLES ON THE EDGE OF THE SAMPLE STREAM.
C
0204      I=IY1-R/DY-1
0205      NYM=2.*(R/DY)+1.
0206      DO 200 J=1,NYM
0207      I=J+M
0208      P=(ABS(R*(R-(YY(I)-Y1)*(YY(I)-Y1))))**.5
0209      ZMIN(J)=Z1-P
0210      200 CONTINUE
0211      WRITE (6,120)
0212      DO 270 J=1,NYM
0213      DO 270 K=1,NYM
0214      L=K+M
0215      ZF(K)=ZMIN(K)+XLONG*(W(L)+10*FNCCL(J)*CAGC(L))/U(L)
0216      IF (YY(L).GE.YD.OR.YY(L).LE.-YD)
C
X      ZF(K)=999.
0217      F=(ABS(R*(R-(YY(I)-Y1)*(YY(L)-Y1))))**.5
0218      ZF2(K)=ZF(K)+2.*P
0219      IF (ZF(K).LE.-ZW)
C
X      WRITE (6,126) FNCCL(J),YY(L)
270      IF (ZF2(K).GE.ZW)
C
X      WRITE (6,126) FNCCL(J),YY(L)
C
C   FIND THE Y-AVERAGED MOBILITY DISTRIBUTIONS AT EACH Z POINT.
C
0221      DO 230 I=1,NZ
0222      DO 210 K=1,NYM
0223      L=K+M
0224      IF (ZZ(I).GE.ZF(K).AND.ZZ(I).LE.ZF2(K))
C
X      AM(K)=FNCCL(J)*U(L)
0225      210 IF (ZZ(I).LT.ZF(K).OR.ZZ(I).GT.ZF2(K))

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	X	AS(K)=J.	337
0226		CALL SIMPS(DY,AM,1,MYH,D)	338
0227		FNBAR(I)=D/USCALI/DEPTH	339
0228	230	CONTINUE	340
	C		341
	C	FIND THE AVERAGE MOBILITY DISTRIBUTION IN EACH SAMPLE COLLECTOR.	342
	C		343
0229		NDIV=NFCCL-1	344
0230	260	J=1,NCCL	345
0231		JINIT=(1-1)*NDIV	346
0232	270	K=1,NCCL	347
0233		I=K+JINIT	348
0234	240	EM(K)=FNBAR(L)	349
0235		CALL SIMPS(DZ,EM,1,NFCCL,C)	350
0236		C=C/ZCCL	351
	C		352
	C	PRINT THE COLLECTOR MOBILITY DISTRIBUTIONS.	353
	C		354
0237		WRITE (6,127) FMOEL(J),I,C	355
0238	260	CONTINUE	356
0239	330	CONTINUE	357
0240	99	STOP	358
0241		END	359

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TEMP

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0001      SUBROUTINE TEMP(MI,NS,FS,FKP,T)
C
C THE TEMP SUBROUTINE EVALUATES THE TEMPERATURE FIELD T = TZERO + FKP*ICONE
C TZERO IS THE ZERO ORDER FIELD- TONE THE FIRST ORDER FIELD. THE MANY GI
C THE TEMPERATURE AT NTEMP POINTS FROM Y=0 TO Y=1
C
0002      DIMENSION T(1000)
C
C CALCULATE ZERO ORDER FIELD
C
0003      FI=3.1415926
0004      FN1=FS**(0.5)
0005      CFN1=CCS(FN1)
0006      IFN1=2.*FN1
0007      EY=1./(NI-1)
0008      DO 2 J=1,NI
0009          Y=EY*(J-1)
0010          X=(1.+Y)*FI/2.
0011          TZERO=(CCS(FN1*Y)-CFN1)/CFN1/FS
C
C CALCULATE FIRST ORDER FIELD
C
0012      SUM=0.
0013      FKP=4.*FS/PI/PI
0014      DO 3 JI=1,NS
0015          FNP=JI*PI/2.
0016          FK1=SIN(FNP)
0017          FC1=TFN1+FNP
0018          FC2=TFN1-FNP
0019          A=SIN(FC1)/4./FC1+SIN(FC2)/4./FC2+FN1/YNE/2.
0020          B=A*FI*FK1
0021          FC1=FN1-FNP
0022          FC2=FN1+FNP
0023          B=SIN(FC1)/2./FC1+SIN(FC2)/2./FC2
0024          E=-E*2.*FI*FK1*CFN1
0025          NC=1-(-1)**JI
0026          C=CFN1*CFN1*NC/JI
0027          FJIS=JI*JI
0028          ASUM=(A+B+C)*FJIS/2./(FJIS-FK3)
0029          ASUM=ASUM*SIN(JI*X)
0030      3      SUB=SUM+ASUM
0031          TONE=-SUM/FS/FS/CFN1/CFN1
0032          ICNE=ICNE*2./FI
0033      2      T(J)=TZERO+FKP*ICNE
0034      RETURN
0035      END

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SIMPS

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0001	SUBROUTINE SIMPS(H,A,N1,N2,AREA)	4060
	C	4070
	C THE SIMPS SUBROUTINE USES SIMPSON'S RULE TO INTEGRATE AN ARRAY	4080
	C	4090
0002	REAL PIDSUM	4100
0003	DIMENSION A(1001)	4110
0004	ENDSUM=0.	4120
0005	MIDSUM=0.	4130
0006	NI=(N2-N1)/2	4140
0007	J=N1-1	4150
0008	DO 1 I=1,NI	4160
0009	J=J+1	4170
0010	ENDSUM=ENDSUM+A(J)	4180
0011	J=J+1	4190
0012	1 MIDSUM=MIDSUM+A(J)	4200
0013	AREA=H*(2.*ENDSUM+4.*MIDSUM-A(N1)+A(N2))/3.	4210
0014	RETURN	4220
0015	END	4230

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