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Wind-Tunnel Tests of the XV-15 Tilt Rotor Aircraft

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NOTATION

Symbol	Computer Notation	Description	
b	SPAN	wing span	32.17 ft
c	CBAR	wing chord	5.25 ft
C_D	CD	drag coefficient	D/qS
C_l	CROLL	rolling-moment coefficient	L/qSb
C_L	CL	lift coefficient	L/qS
C_m	CM	pitching-moment coefficient	M/qSc
C_n	CN	yawing-moment coefficient	N/qSc
C_p	CP	power coefficient	$550HP/\pi R^2\rho(\Omega R)^3$
C_y	CY	side force coefficient	Y/qS
D	DRAG[1]	drag	lb
fe	FE	flat plate drag area	$C_D S$ ft^2
HP_{mast}	HP	rotor shaft power	hp
L	LIFT	lift	lb
L	ROLL	rolling moment	lb-ft
M	PITCH	pitching moment	lb-ft
N	RPM	rotor rpm	rpm
N	YAW	yawing moment	lb-ft
q	QPSF	dynamic pressure	$(1/2)\rho V^2$, psf
Q_M	R MAST Q L MAST Q	mast torque (R - right) (L - left)	in.-lb
R	R	rotor radius	ft
S	AREA	wing area	sq ft
V	VFPS	velocity	fps

[1]U denotes uncorrected values.

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Symbol	Computer Notation	Description	
V_K	VKTS	velocity	knots
X	PED POS	pedal position	% (neutral = 50%)
	LONG STK	longitudinal stick	"
	LAT STK	lateral stick	"
	POW LEV	power lever position	"
	MTUN	tunnel mach. no.	
	TEMP	tunnel temperature	°F
	VSNDKTS	speed of sound	knots
Y	SIDE	side force	lb
α	ALFS,C	fuselage angle of attack	deg
i_N	INA	pylon angle[2]	deg
μ	V/OR	advance ratio	$V/\Omega R$
ΩR	OMEG*R	rotor tip speed	fps
ρ	RHO	air density	lb sec^2/ft^4
σ	SIGMA	rotor solidity ratio[3]	0.089
σ'	ROHR,6	density ratio	$\rho/\rho o$
ψ	PSI	yaw angle (positive nose right)	deg
δ_f		flap deflection	deg

[2]relative to fuselage (0° airplane, 90° helicopter)

[3]blade area/rotor disc area

WIND-TUNNEL TESTS OF THE XV-15 TILT ROTOR AIRCRAFT

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SUMMARY

The XV-15 Tilt Rotor Research Aircraft was tested in the Ames 40- by 80-Foot Wind Tunnel for preliminary evaluation of aerodynamic and aeroelastic characteristics prior to flight. The tests were undertaken to investigate the aircraft performance, stability, control and structural loads for flight modes from helicopter through transition and airplane mode up to the tunnel capability of 170 knots. Results from these tests are presented.

INTRODUCTION

The joint NASA/Army XV-15 Tilt Rotor Research Aircraft Project involves design, fabrication, and flight test of two essentially identical aircraft (fig. 1). The overall plan to implement this program is documented in references 1 and 2. A Test and Evaluation Plan, reference 3, outlines the tests to ensure that the XV-15 aircraft will meet the requirements of the Program Plan and the contract Model Specification and Statement of Work. As part of this plan, one of the aircraft was tested in the Ames 40- by 80-Foot Wind Tunnel (fig. 2).

Prior to the wind-tunnel tests, the operation of the aircraft systems was evaluated on a ground tiedown stand at the contractor's facility (fig. 3). After completing about 40 hr of test time, a brief flight evaluation of the aircraft hover characteristics was made (fig. 4). Three hours of flight time in helicopter mode were accumulated from May 3 to 31, 1977. Flight was limited to 40 knots forward and 100 foot altitude. The ground tiedown tests were then resumed for a total of 130 hr accumulated time. The last 5 hr were with systems installed for remote operation in the wind tunnel. This served not only to check the functional operation of the system but also to train personnel in its operation.

Following completion of the remote control checkout on March 11, 1978, the aircraft was prepared for shipment to Ames for the wind-tunnel tests. This involved removal of the wing from the fuselage and mounting it in a shipping cradle. The components were then airlifted on March 23, 1978 to

Ames, reassembled, and the aircraft installed in the wind tunnel on May 4, 1978 (figs. 5 to 8). Testing was conducted to June 23 and consisted of 51 runs in 54 hr of wind-on time. Of this, 19 hr were with rotors on.

The purpose of the wind-tunnel test was to provide an initial assessment of the aerodynamic and aeroelastic characteristics and structural loads within the aircraft and tunnel operating envelope. The tests also served to verify the functional operation of the aircraft systems and on-board instrumentation in a simulated flight environment.

A brief summary of the results of these tests along with a computer print-out of the wind-tunnel scale data is presented in this report. Additional data and analysis including structural loads and structural dynamics are presented in reference 4.

DESCRIPTION OF THE AIRCRAFT

The XV-15 configuration is based on the Bell Helicopter model 301 design. Two aircraft were built and one of these was tested in the wind tunnel. A detailed description of the aircraft and its systems is given in reference 5. Pertinent geometry is shown in figure 1 and table I.

In order that the aircraft could be mounted and operated in the wind tunnel, modifications were incorporated during construction of the ship's structure and hydraulic, fuel, electrical and control systems. For the wind-tunnel tests, the aircraft was provided with remote operation of engine and flight controls and an external supply of fuel, hydraulics, and electrical power. Actuators for remote operation of the flight controls were installed in the aircraft systems as shown in figure 9. The aircraft electrical and hydraulic systems are normally powered from engine driven generators and transmission driven pumps. To provide for operation in the wind tunnel when engines were not running, the D.C. electrical and one of the hydraulic systems (PC-2) were connected to a tunnel source (ref. 6 and fig. 10). For the tests with engines operating, the aircraft systems were used. Prior to tunnel entry, the ship's fuel tanks were purged and inerted with nitrogen and the fuel lines disconnected and capped (ref. 7 and fig. 11). In the tunnel, the fuel lines to the engines were connected directly to the tunnel supply, thus bypassing the aircraft's fuel tank systems.

For tests with rotors off, the blades with pitch links were removed from the rotor hub spindles (fig. 8). To prevent engine compressor windmilling with power off and tunnel wind on, the engine inlet ducts and exhaust were sealed (fig. 12).

During the test, various aircraft configuration changes were made for aerodynamic improvements or to correct problems (vibration, low C_L max and flap effectiveness). These configuration modifications are shown in figure 13 and table II and included pylon strakes, fences, vortex generators, flap tabs, and structural supports.

Because of an interference between the landing gear doors and the tunnel support struts, the entire test was run with the gear retracted and the gear doors closed.

TUNNEL INSTALLATION

The installation of the aircraft in the wind tunnel is shown in figure 14. The aircraft was supported on a system of 3 struts (ref. 8). To accept the loads imposed by attachment to the support struts, additional structure was added to the wing and fuselage in the attachment area during initial construction. In selecting the strut arrangement, consideration was given to dynamic characteristics and rotor/tunnel clearances (ref. 9). To verify predicted mode shapes, frequency and damping of the aircraft as mounted in the tunnel, a dynamic shake test of the combined aircraft, support system and tunnel balance was made after installation and prior to wind on testing. The shake tests were conducted with rotor blades off and with weights attached to the hub spindles to simulate the blade mass. Excitation was applied to one wing tip using an electrohydraulic actuator (fig. 15). Accelerometers and strain gages were used to measure the applied force and structural response. The shake tests were conducted using the procedures and data analysis system described in reference 10. No resonance problems at operating conditions were indicated. Details of the results are presented in reference 4.

Remote operation of the aircraft in the wind tunnel was controlled from a console in the control room (fig. 16). The console contained the controls for operation of the actuators in the aircraft for remote operation of engine and flight control systems. Instruments for monitoring the conditions of the various aircraft systems were removed from the cockpit and installed in the console.

Fuel and hydraulic lines from the tunnel systems were routed inside the right strut and connected to the aircraft systems inside the wing aft of the rear spar (fig. 17). Electrical and instrumentation leads were routed inside the left strut. Because of space requirements, these leads were routed externally on the wing lower surface from the top of the strut to connect with the ship's system inside the fuselage (fig. 18).

To assure safe operation of the tunnel test, an analysis was made to evaluate the safety provisions in the aircraft and for test operation procedures. The results are presented in reference 11. As part of this safety evaluation and prior to the tunnel tests, crew training was conducted during the latter part of the ground tiedown tests at the contractor's facility with the remote systems installed.

In addition, a simulation of the test operation to evaluate significant failure and recovery modes was conducted at Ames using the FSAA flight simulation program. This failure mode evaluation used the aircraft math model that was used for the flight simulation without the automatic flight control systems, FFS, and SCAS, and did not require the use of the FSAA cab. Inputs to the math model were adjusted to correspond to the aircraft as mounted in the tunnel. Remote operation in a simulated tunnel environment and emergency operating procedures were evaluated. The only significant failure identified that could cause a dangerous condition (high blade flapping and high loads) was a simultaneous dual engine failure at nacelle incidences above 85° (i.e., helicopter mode). Recovery from this failure was to reduce nacelle incidence within 5 sec of the failure. Complete conversion from 95° to 0° can be accomplished in 11 sec. If nacelle incidence is 85° or less at the time of the dual power failure, no corrective action was required.

DATA ACQUISITION AND REDUCTION

The wind-tunnel data acquisition system is a computer-based system that can be operated in either an on-line mode or stand alone with batch processing mode. The components of this system are described in reference 8. The components used for this test and the interface with the aircraft systems is shown in figure 19. The equipment is located in the wind-tunnel computer and control room (fig. 16(a)). Data acquired during the test included static aerodynamic forces, structural loads, and the status of the aircraft systems.

The static forces on the aircraft were measured by the wind-tunnel six-component balance system. Forces and moments were computed about the wind axes and the moment center shown in figure 20. No corrections were applied for support strut tares or tunnel-wall interference. Dimensional data used for reduction to coefficients are given in table I. Angle of attack and yaw are referenced to the fuselage reference line. Some of the data for setting and monitoring test conditions were displayed on a CRT (fig. 21).

Test operations involved monitoring loads in critical structural components and monitoring the condition of the various aircraft systems including the hydraulic, electrical, flight controls, and propulsion systems. Data acquisition, reduction, and display utilized the aircraft on-board instrumentation and the tunnel instrumentation data reduction and display systems.

The aircraft on-board instrumentation system is described in reference 12 and consists of sensors, signal conditioning, encoding (to Pulse-Code-Modulation (PCM) digital format) and recording. During the tunnel test, the aircraft's data system tape recorder was installed in the control room to provide access during test operations. The wind-tunnel data acquisition systems are described in reference 8 and have monitoring, reduction, analysis, display, and recording capabilities. The on-board research instrumentation and the interface with the tunnel systems are listed in table III and shown in figure 19.

Structural loads were displayed for monitoring on the wind-tunnel Peak Detector System (PDS), Cathode Ray Oscilloscope (CRO), brush recorders and a loads panel on the control console. The PDS computes peak-to-peak amplitude of selected dynamic signals in percent of full scale monitoring limit and displays the amplitude on a 50-channel bar graph display on a 20-in. CRT. Each channel was set to alarm when monitoring limits were exceeded. The CRO displayed, on an 8-in. CRT, the dynamic structural loads from one blade and hub spindle as X-Y pairs (beam vs. chord) and the corresponding pitch link load to allow an on-line assessment of loads with respect to allowables or critical limits. Some selected dynamic loads were displayed and monitored on brush recorders. The remote control console had a panel for display of loads in 12 structural elements for the information of the console operators (fig. 16(c)).

TESTS AND RESULTS

Testing followed the Test Plan of reference 13. A log of the runs completed is given in table IV. The conversion corridor area (the aircraft configuration between the helicopter and the airplane flight modes covered with rotors on) is shown in figure 22. A computer printout of the tunnel-scale data is given in table V. Some of this data is plotted and is shown in figures 23 to 31.

Some of the significant observations are discussed below.

Lift Coefficient

The measured lift characteristics of the XV-15 at all nacelle angles demonstrated an apparent premature flow separation resulting in a reduction of achievable lift at high angles of attack. Inspection of the wing airfoil contours indicate that leading edge out-of-contour and surface irregularities may have contributed to this problem. The addition of vortex generators at the upper-surface quarter-chord location along the full wing span (fig. 13) eliminated the premature stall and retained attached flow up to an angle of attack of 15°. This improvement is shown in figure 24(a) for the airplane mode configuration. The addition of vortex generators at the 60% chord ahead of the flaps had no significant effect on lift up to angles of attack of 10° (fig. 24(b)).

Flap Effectiveness

The lift increment due to flap deflection was improved by the addition of a flap tab. This 3/4-in. tab runs the full span of the wing trailing edge and it projects downward approximately normal to the wing chord (fig. 13). The measured lift, drag, and moment data presented in figure 25(a) illustrates the lift increase due to the tab and its impact on drag. At low levels of lift coefficient and flap angles of 0° and 20° there is no effect on drag. As illustrated in figure 25(b), the flap tab provides a significant increase of the incremental lift due to flap deflection at 0° aircraft angle of attack. At high lift coefficients and at a 40° flap angle, a drag increase is apparent, as shown on the C_D and L/D curves of figure 25(a).

Autorotation

Autorotational capability was demonstrated at an 80-knot airspeed with a 95° nacelle angle and 40° flap angle aircraft configuration. The rotor RPM governor was engaged and several rotor speeds were tested. At the zero mast-torque autorotational condition shown in figure 27(g), the minimum rate of descent is computed to be approximately 2450 ft/min at 76% RPM.

Yaw Characteristics

The effect of aircraft yaw on the force and moment coefficients is shown in figures 23(b) and 28. Of note is the yawing coefficient, C_n, which demonstrates a linear variation with yaw angle. A reduction of the slope of C_n around the zero yaw angle measured in model tests of the original single vertical fin configuration of the tilt rotor research aircraft caused concern about its directional stability characteristics. No problem in this area was detected with the full-scale H-tail XV-15.

Drag

A determination of the drag of the XV-15 in a free-air state was not obtained because of the difficulty in establishing the tares and interference corrections with precision and confidence. The measured drag, however, indicated levels greater than expected based on the results of prior small scale tests. A summary of the rotors-off airplane mode drag, as measured by the wind-tunnel balance, is shown in figure 30. The configurations of the XV-15 on the three strut support system, in order of decreasing drag, are: vortex generators on; vortex generators off; and a clean (gaps taped, remote control cables and lines removed) configuration.

An indication of the drag coefficient difference between the three strut and a single strut support system is also presented in figure 30 based on a 1/5 scale model test. It appears that the pretest prediction would underestimate the free-air XV-15 minimum drag by as much as 20%. Comparisons of the clean aircraft configuration data indicate that the anticipated drag reduction at full scale Reynolds numbers is not occurring.

Airflow

The placement of tufts on the aircraft during the wind-tunnel tests revealed several areas of disturbed boundary-layer flow. For all flight modes, the upper-aft portion of the landing gear pod and the fuselage surface aft of the pod showed turbulent flow. Without vortex generators, the upper surface of the wing aft of 60% chord indicated separation when the flaps were deflected. Some separation was also noted on the upper fuselage just aft of the wing above the trailing edge fillet. In helicopter and conversion flight, the upper, or aft surface of the nacelles, showed severely separated flow. (Note that the nacelle is roughly oval in shape and is canted outboard at helicopter and conversion nacelle angles.) During portions of the conversion envelope (particularly when high oscillatory empennage loads were present), the vertical fin tufts reflected turbulent flow conditions.

Systems Operation

Throughout the wind-tunnel test period, all aircraft components and systems operated well and within acceptable tolerances with the exception of the following items:

Pylon downstop— A failure of the right-hand pylon downstop bracket occurred. This component provides the hard-point that the pylon engages in the aircraft mode to increase the wing/pylon stiffness for aeroelastic stability. The failure was subsequently determined to have resulted from overstress cycle fatigue. The crack appeared to have started prior to the wind-tunnel test. The left-hand downstop also was cracked in the same location. As a result of this failure, only three runs had to be deleted from the planned run schedule. A limited amount of airplane mode testing was conducted by raising the pylons to just off the downstop. Following the wind-tunnel test, a redesign of the downstop bracket and a modification of its rigging/preloading procedure was initiated.

Nose boom vibration— During operation in airplane mode at speeds greater than 150 knots, the nose boom and the pitch and yaw vanes on the nose boom YAPS head vibrated excessively. Between runs 6 and 24, the nose boom YAPS head was removed. The nose boom was then stiffened by adding support wires for the remainder of the test.

Antenna vibrations— Excessive vibrations of the aft-fuselage mounted VOR-LOC antennas occurred during attitude sweeps in airplane mode flaps down. Inspection disclosed a structural failure at the attach point. The vibration was felt to be the result of wing flow separation. The antennas were removed for the remainder of the test.

Flap drive— During a run with flaps down, the flaps could not be extended beyond about 30° while operating at 160 knots. The failure was found to be due to a loose wire.

Engine oil venting— Seepage from the engine oil scavenge lines occurred during initial runs with rotors off while operating in airplane mode at airspeeds greater than 100 knots. The problem was corrected by scarfing the end of the tube. The modification was satisfactory for the remainder of the test up to the maximum test airspeeds.

Empennage loads— Oscillatory load limits were encountered on the empennage structure during helicopter and conversion mode operations. In the helicopter mode, at a nacelle incidence of 90°, the horizontal spar attach lug tension loads increased with increasing airspeed up to approximately 40 knots (as previously indicated by flight test of aircraft no. 1). Above 40 knots, the load amplitude decreased rapidly with airspeed (fig. 31(a)). The addition of a preload strap reduced the loads on the lugs to well below the design limits. The excitation for these loads is believed to be the tip vortices shed from the inboard edge of the rotor disk. These vortices, which are generated downward in the hover condition, are swept aft as forward flight is initiated, providing a strong oscillatory flow at the empennage at about 40 knots. At higher speeds, the rotor tip vortices pass clear of the empennage.

At the low speed end of the conversion envelope, for nacelle angles around 60°, high vibratory loads appeared on the empennage structure, with the horizontal spar experiencing loads above the infinite life design limits. The placement of struts (shown in figs. 13(a) and (d)) reduced the loads in the horizontal tail structure but resulted in the growth of vertical fin spar and attach fitting loads to beyond its design limits. A series of tests after run 23 examined the effect of various aerodynamic modifications such as wing fences, nacelle strakes, and vortex generators (fig. 13) on the tail loads. Some of these results are shown in figures 31(b) and 31(c). No aerodynamic modification solution was found to be sufficiently effective and suitable for all flight-mode (hover through airplane) conditions. Long streamers attached to the inboard side of the nacelle at the wing/nacelle junction showed that a strong vortex rolls over the forward portion of the wing tip and is swept inboard to the lower half of the vertical fins at 60° nacelle incidence. Subsequent flight tests showed this problem to be less severe than indicated by wind-tunnel tests.

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12. Instrumentation and Data Acquisition Manual. Bell Helicopter Textron Rept. 301-099-003B, June 1974.

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TABLE I. AIRCRAFT DIMENSIONS
(English Units)

(a) General

	Wing	Horizontal Tail	Vertical Tail
Area, sq ft	181.0[a]	50.25	50.5[b]
Span, ft	32.17[c]	12.83	7.68
Chord, ft	5.25	3.92	4.09/2.40[d]
$\bar{c}$, ft[e]	5.25	3.92	3.72
Aspect ratio	6.12	3.27	2.33
Incidence, deg	3.0	0	--
Dihedral, deg	2.0	0	--
Sweep $\frac{c}{4}$, deg	-6.5	0	31.6[f]
Section (NACA)	64A223 (Mod)	64A015	0009
Tail length, ft	--	22.4	23.2

(b) Movable Aerodynamic Surfaces

	Flap	Flaperon	Elevator	Rudder
Area aft hinge, sq ft	11.0[b]	20.2[b]	13.0	7.5
Span, ft	4.25[g]	7.86[g]	11.0	4.66
Chord aft hinge, %	25	25	30	25
Deflection, deg	75 max	-23.8[h] +13.8	±20	±20
Control travel for total surface deflection, in.	--	9.6	9.6	5.0

(c) Rotor

Number of blades per rotor	3
Diameter, ft	25.0
Disc area per rotor, sq ft	491
Blade chord, in	14.0
Blade area per rotor, sq ft	43.75
Solidity	0.089
Blade twist, deg	
Aerodynamic	45.
Geometric	40.9
Hub precone angle, deg	2.5
δ_3, deg	-15.0
Flapping design clearance, deg	±12
Blade Lock number	3.83
Rotor rpm/tip speed	
Helicopter	565 rpm/740 fps
Airplane	458 rpm/600 fps

a To BL 207

b Total both panels

c Between rotor centerlines at tilt axis

d Root/tip

e Mean aerodynamic chord

f Upper

g One side along hinge

h Measured from flap. Also flaperon droops .625°/deg. flap

TABLE I.- Concluded.
(Metric units)

(a) General

	Wing	Horizontal Tail	Vertical Tail
Area, sq m	16.81[a]	4.67	4.69[b]
Span, m	9.80[c]	3.91	2.34
Chord, m	1.60	1.19	1.25/.73[d]
$\bar{c}$, m[e]	1.60	1.19	1.13
Aspect ratio	6.12	3.27	2.33
Incidence, deg	3.0	0	--
Dihedral, deg	2.0	0	--
Sweep $\frac{c}{4}$, deg	-6.5	0	31.6[f]
Section (NACA)	64A223 (mod)	64A015	0009
Tail Length, m	--	6.83	7.07

(b) Movable Aerodynamic Surfaces

	Flap	Flaperon	Elevator	Rudder
Area aft hinge, sq m	1.02[b]	1.88[b]	1.21	.70[b]
Span, m	1.29[g]	2.40[g]	3.35	1.42
Chord aft hinge, %	25	25	30	25
Deflection, deg	75 (max)	-23.8[h] +13.8	±20	±20
Control travel for total surface deflection, cm	--	24.4	24.4	12.7

(c) Rotor

Number of blades per rotor	3
Diameter, m	7.62
Disk area per rotor, sq m	45.6
Blade chord, cm	35.6
Blade area per rotor, sq m	4.06
Solidity	.089
Blade Twist, deg	
Aerodynamic	45.
Geometric	40.9
Hub precone angle, deg	2.5
δ_3, deg	-15
Flapping design clearance, deg	±12
Blade Lock number	3.83
Rotor rpm/tip speed	
Helicopter	565 rpm/226 mps
Airplane	458 rpm/183 mps

TABLE II. AIRCRAFT CONFIGURATION LOG

			(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)	(10)	(11)	(12)	(13)	(14)	(15)	(16)	Ref fig. 13(a)
				Vortex Generators							Pylon Strakes						Fences		Remarks
Run No.	Rotors	Nose boom YAPS head	Horizontal tail struts	1"x2½" @20° ∠ on .25c	1"x1¼" @20° ∠ on .6c	3"x5" @20° ∠ on .6c	.6"x1½" @15° ∠ on .25c	Flap tab full span T.E.	6" high full length 45° ob'd	18" high splitter top side	3"x34" lower ob'd	3"x16" outboard	3"x8" outboard	3"x18" outboard	15"x52" 45° ob'd	Engine inlet fairing	4"x15" 10" ib'd	12" high flap. to 8" on bot.	
1-5	OFF	ON																	
6		ON/OFF																	Removed during Run 6
7-8		OFF																	
9-10				ON															
11																			Heli. doors off $ 3-5
12-13	OFF		ON																
14-19	ON																		
20-21					ON	ON													
22-23		OFF	ON																
24-25		ON	OFF						ON										
26	ON		ON																
27-29	OFF																		
30						ON		ON	ON										
31						OFF			OFF										Top cowl doors off
32																	ON		Top cowl doors on
33																	OFF		Sealed pylon/wing gap
34										ON	ON								Gap seals off
35										ON	ON	ON	ON						Pylon i.b'd opening closed Cowl fairing off
36-39										OFF	OFF	OFF	OFF						Var. cowl fairings & gap seals
40-41			ON											ON					
42			OFF																
43			ON		ON											ON			
44				ON	OFF														
45		ON		OFF			ON	ON						ON					
46		OFF						OFF						OFF					
47																			Taped gaps
48															ON				
49															extended 15			ON	
50							ON								ON			OFF	
51	OFF	OFF	ON	OFF	OFF	OFF	OFF	OFF	OFF	OFF	OFF	OFF	OFF	OFF	OFF	ON	OFF	OFF	

FOLDOUT FRAME 1

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FOLDOUT FRAME 2

TABLE III.- INSTRUMENTED PARAMETERS

Measurement no.	Description	Input TSP	PCM word	CRO	HS	TCS	DAS	p-p	DPM	OBS	DAC	
B122	Right bld beam 52.5		B1-10	2	1		29	1			A6	Load panel
B132	Left bld beam 52.5		B1-8		2		30	2			A4	Load panel
B123	Right bld chord 52.5		B1-20	6	3			3			A10	
B133	Left bld chord 52.5		B1-9		4		31	4			A5	
F103	Right pitch link		B1-45	3	5		28	5			A24	Load panel
F060	Left pitch link		B1-21		6		32	6			A11	Load panel
B165	Right conv spindle beam		B1-57		7			7			A27	Load panel
B190	Left conv spindle beam		B2-54		8			8			A29	Load panel
B166	Right conv spindle chord		B2-57		9			9			A30	
B191	Left conv spindle chord		B1-55		10			10			A26	
F611	Right conv actuator load		B1-7		11			11			A3	
F638	Left conv actuator load		A1-5		12			12			C1	
B112	Right yoke beam 9.0		B1-48	1	13			13			A25	
B114	Left yoke beam 9.0		B1-15		14			14			A7	
B113	Right yoke chord 9.0		B1-34	5	15			15			A18	Load panel
B115	Left yoke chord 9.0		B1-16		16			16			A8	Load panel
B109	Right perp mast bending		A1-21		17			17			C4	Load panel

TABLE III.- Continued.

Measurement no.	Description	Input TSP	PCM word	CRO	HS	TCS	DAS	p-p	DPM	OBS	DAC	
B141	Left perp mast bending		B1-18		18			18			A9	Load panel
M107	Right mast torque		B1-6		19		14	19	3		A2	Load panel
M143	Left mast torque		B1-5		20		15	20	2		A1	Load panel
D156	Right F/A flapping	T104			21		22	21				AC cons.
D181	Left F/A flapping	T122			22		23	22				AC cons.
D157	Right lat. flapping	T101			23		24	23				AC cons.
D182	Left lat. flapping	T119			24		25	24				AC cons.
B656	Right wing beam 22	T107, 108, 109			25	1	4	25				
B626	Left wing beam 22	T125, 126, 127			26	2	5	26				
B655	Right wing chord 22	T110, 111, 112			27	3	6	27				
B625	Left wing chord 22	T128, 129, 130			28	4	7	28				
M657	Right wing torsion 22	T113, 114, 115			29	5	8	29				
M627	Left wing torsion 22	T131, 132, 133			30	6	9	30				
E747	Flaperon exciter	T137			31		2					
E746	Collective exciter	T140			32		3					
A150	Right pylon F/A accel.		B1-37		33		16				A19	

TABLE III.- Continued.

Measurement no.	Description	Input TSP	PCM word	CRO	HS	TCS	DAS	p-p	DPM	OBS	DAC	
A175	Left pylon F/A accel.		B1-30		34		17				A15	
A151	Right pylon lat. accel.		B1-38		35		18				A20	
A176	Left pylon lat. accel.		B1-31		36		19				A16	
A152	Right pylon vert. accel.		B1-39		37		20				A21	
A177	Left pylon vert. accel.		B1-32		38		21				A17	
RSFAA	Right strut F/A accel.	T004, 005, 006			39	7	10					
LSFAA	Left strut F/A accel.	T010, 011, 012			40	8	11					
RSLAA	Right strut lat. accel.	T001, 002, 003			41	9	12					
LSLAA	Left strut lat. accel.	T007, 008, 009			42	10	13					
B262	Right horiz. beam 8.0		B1-58		43		27	31			A28	
B259	Left horiz. beam 8.0		B2-60		44		26	32			A32	
B263	Right horiz. chord 8.0		B2-58		45						A31	
F052	Right swash plate driver		B1-44		46			33			A23	
F142	Left swash plate driver		B1-24		47			34			A12	
F162	Right F/A cyc act load		B1-40		48			35			A22	
F187	Left F/A cyc act load		B1-27		49			36			A13	
F163	Right lat. cyc act load		A1-29		50			37			C5	

TABLE III.- Continued.

Measurement no.	Description	Input TSP	PCM word	CRO	HS	TCS	DAS	p-p	DPM	OBS	DAC	
F188	Left lat. cyc act load		B1-28		51			38			A14	
F164	Right collective act load		B1-19		52			39			B1	
F189	Left collective act load		B1-29		53			40			B2	
B120	Right blade beam 22.8		A1-12		54			41			C2	
B130	Left blade beam 22.8		B1-43		55			42			B5	
D158	Right collective pitch		B4-82		56						B13	AC cons.
D183	Left collective pitch		B3-79		57						B10	
D159	Right F/A cyclic		B5-82		58						B15	
D184	Left F/A cyclic		B4-79		59						B12	
D160	Right lat. cyclic		A2-72		60						C6	AC cons.
M266	Right horiz. torsion 8.0		B1-59								B6	BNC 1
B124	Right blade beam 75		B1-35								B3	BNC 2
B125	Left blade chord 75		B1-36								B4	BNC 3
B126	Right blade beam 112		A1-16								C3	BNC 4
D185	Left lat. cyc		B5-79								B14	AC cons.
	Shaker excitation											AC cons.
	Az right											
	Az left											
	NPR right											CTR 1
	NPR left											CTR 2

TABLE III.- Concluded.

Measurement no.	Description	Input TSP	PCM word	CRO	HS	TCS	DAS	p-p	DPM	OBS	DAC	
D021	F/A stick pos		B1-78						1		B7	
D022	Lat. stick pos		B2-78						7		B8	
D023	Power lever pos		B3-78						5		B9	
D024	Pedal pos		B4-78						6		B11	

TABLE IV.- RUN LOG
(Rotors off)

Run no.	Pt no.	i_N	V_{knots}	δ_F	Cntr	Sweep	α trim	Run time	Remarks
3	2-12	90	80	40	1706-1708	α: -12 to +14	-	1:53	Clean A/C
	13-19	90	80	40	1709-1711	ψ: -4 to +16	-8		
	20-25	90	80	40	1712-1718	$\Delta X_{long}, \Delta X_{lat}, \Delta X_{ped}$	-8		
5	2-12	90	60	75	1723-1727	α: -12 to +14	-	1:30	
	13-19	90	60	75	1728-1730	ψ: -4 to +16	-8		
	20-21	90	60	75	1731-1732	ΔX_{lat}	-8		
6	5-10	0_L	160	0	1742-1743	α: -4 to +8	-	2:25	Removed nose
	11-17	0_L	122	0	1744-1746	α: -4 to +12	-		boom
	18-23	0_u	160	40	1747-1749	α: -4 to +8	-		Removed VHF
	24-30	0_u	110	40	1750-1751	α: -4 to +12	-		antennas
7	4-7	0_L	-	0	1755-1757	V: 100 to 178	0	1:26	Dyn. stability
8	3-4,7	90	-	40	1761-1763	V: 80 to 118	0	1:40	Dyn. stability
	5-6,10	0_u	-	40	1767	V: 111 to 170	0		Dyn. stability
	8-9	30	-	40	1765-1766	V: 122 to 170	0		Dyn. stability
9	3-14	0_L	80	0	1772-1780	α: -4 to 16	-	2:15	VGs on
	15-25	0_L	80	40	1781-1788	α: -4 to +14	-		
	26-31	-	80	40	1789-1794	i_N: 0 to 90	0		
	32-41	0_L	110	40	1795-1801	α: -4 to +12	-		
	42	0_L	110	0	-	-	0		
10	4-11	0_L	160	0	1803-1807	α: -4 to +8	-	1:08	
	12-15	30	140	40	1808-1810	α: -4 to +2	-		
	16-19	75	100	40	1811-1814	α: -4 to +1	-		
	20-22	60	120	40	1815-1817	α: -4 to -1	-		
11	3-5	75	100	40	1818-1820	α: -4 to 0	-	1:32	Removed heli-
	6-10	0_L	160	0		ψ: -4 to +8	+2		copter doors
	11-16	0_L	160	0		$\Delta X_{long}, \Delta X_{lat}, \Delta X_{ped}$	+2		
	17-19	0_L	125	0		ΔX_{lat}	+8		
	20-22	0_L	125	0		ΔX_{lat}	+12		
	23-27	0_L	160	40		ψ: -4 to +8	0		
	28-33	0_L	160	40		$\Delta X_{long}, \Delta X_{lat}, \Delta X_{ped}$	0		
	34-36	0_L	110	40		ΔX_{lat}	+8		
12	3-13	90	80	40	1856-1864	α: -12 to +14	-	1:00	Tail strut on
	14-18	75	100	40	1865-1867	α: -4 to +4	-		
13	3-7	75	100	20	1868-1872	α: -4 to +4	-	:45	
	8-10	75	100	0	1873-1875	α: +4 to +8	-		
	10-13	-	100	0	1876-1878	i_N: 75 to 0_L	+8		

TABLE IV.- Continued.
(Rotors on)

Run no.	Pt no.	i_N	V_{knots}	δ_F	Cntr	Sweep	α trim	Run time	Remarks
-	-	90	-	40		Track/balance	0	:10	94% N_R
14	3-4	90	-	40		Track/balance	0	:55	
	-	90	-	40		-	0	:02	Xms chip light
15	7	90	40	40	1901		0	1:45	
	8-9	80	-	40	1903-1904	V_{knots}: 40-80	0		
	10-18	-	80	40	1906-1914	i_N: 80-10	0		
	19-22	-	100	40	1915-1918	i_N: 10-0	0		
	23-24	0_L	100	0	1919-1923	Exciter	0		Dyn. stab. 76%
16	-	0_L	120	0		Exciter	0	:20	Oil drain leak
17	3-6	0_L	-	0	1932-1943	V_{knots}: 140-180	0	2:09	Dyn. stability
	7-8	0_L	180	0	1946-1948	Q_m: 60-80%	0		Dyn. stability
18	4-7	0_u	-	40	1952-1956	V_{knots}: 110-170	0	2.25	XMS chip
	8-9	30	-	40	1957-1969	V_{knots}: 150-170	-4		Dyn. stability
	10-16	60	-	40	1970-1977	V_{knots}: 130-151	-7		Dyn. stability
	17	90	80	40	1980	-	-7		Dyn. stability
20	8-15	90	-	40	1989-1997	V_{knots}: 80-100	-8	2:45	Dyn. stability
	16-20	90	80	40	1998-2002	α: -12 to -4	-		
	21-28	90	80	40	2003-2009	ψ: -4 to +16	-8		
	29-34	90	80	40	2010-2014	$\Delta X_{long}, \Delta X_{lat}, \Delta X_{ped}$	-8		
	35-40	90	-	40	2015-2020	V_{knots}: 95-123	-8 to -12		Dyn. stability
	41-44	75	140	40	2021-2024	Q_m: 45-70%	-8		
	45-46	45	160	40	2025-2026				
21	3-10	60	80	40	2027-2034	α: -8 to +7	-	1:00	
	11-16	60	120	40	2036-2041	α: -5 to +3	-		
22	3-5	90	-	40	2052-2057	V_{knots}: 40-80	0	2:08	
	6-10	60	120	40	2058-2062	ψ: -4 to +8	-1		
	11-16	60	120	40	2063-2068	$\Delta X_{long}, \Delta X_{lat}, \Delta X_{ped}$	-1		
	18-28	95	80	40	2070-2082	α: 0 to +12	-		76% N_R Autorotation
	29-38	95	80	40	2083-2094	α: 0 to +12	-		94% N_R Autorotation
	39	95	80	40	2095		12		102% N_R Autorotation
	-	90	60	40	2097-2098		0		Start/stop
23	3-9	0_u	110	40	2099-2104	α: 0 to +12	-	1:20	94% N_R
	10-14	0_u	110	40	2105-2109	ψ: -4 to +8	8		
	15-21	0_u	110	40	2110-2115	$\Delta X_{long}, \Delta X_{lat}, \Delta X_{ped}$	8		
	22-27	0_u	160	40	2116-2120	α: -1 to +3	-		Down stop broke
24	3-4	90	-	40	2127-2128	V_{knots}: 60-80	-5, -8	2:05	Struts off/ 6" strake
	5-9	-	80	40	2129-2133	i_N: 90-30	0		
	10-14	60	80	40	2134-2138	α: -4 to +3	-		
	15-16	60	100	40	2140-2141	α: -2 to 0	-		
	17-20	60	120	40	2142-2145	α: -4 to 0	-		
	21	60	120	0	2146	-	0		
	22-24	75	100	40	2147-2149	α: -5 to -2	-		
	25-30	75	100	40	2150-2155	ψ: -4 to +12	-2		
	31-36	75	100	40	2157-2162	$\Delta X_{long}, \Delta X_{lat}, \Delta X_{ped}$	-2		
	37-41	30	140	40	2163-2167	α: -5 to +3	-		
26	3-5	90	40	40	2173-2175	α: -2, -5, -8	-	1:57	Struts on
	6-10	30	140	40	2176-2180	ψ: -4 to +8	-1		
	11-16	30	140	40	2181-2186	$\Delta X_{long}, \Delta X_{lat}, \Delta X_{ped}$	-1		
	17-20	45	100	40	2187-2190	α: 4 to +9	-1		
	21-28	45	100	40	2191-2197	$\Delta X_{long}, \Delta X_{lat}, \Delta X_{ped}$	+8		
	-	-	80	40	-	i_N: 45-80	-1		
	29-33	80	80	40	2198-2203	α: -5 to +3	-		
	34-40	80	80	40	2204-2209	$\Delta X_{long}, \Delta X_{lat}, \Delta X_{ped}$	-1		
	41-46	80	80	40	2210-2215	ψ: -4 to +12	-1		

TABLE IV.- Continued.
(Rotors off)

Run no.	Pt no.	i_N	V_{knots}	δ_F	Cntr	Sweep	α trim	Run time	Remarks
29	3-9	60	120	40	2218-2224	α: -4 to +6	-	1:45	Strut, 6" strake
	10-16	45	120	40	2225-2231	α: -4 to +8	-		Nose boom on
	17-18	45	120	40	2232-2233	ΔX_{lat}	8		
	19-27	0_u	120	40	2237-2245	α: -4 to 11	-		
	28-29	0_u	120	40	2247-2248	ΔX_{lat}	8		
	30-37	0_u	120	20	2249-2256	α: -4 to +10	-		
	38-45	0_u	120	0	2257-2264	α: -4 to +10	-		
30	3-10	0_u	120	20	2265-2272	α: -4 to +10	-	1:00	Tab on flap (3/4)
	11-17	45	120	20	2273-2279	α: -4 to +8°	-		
	18-19	45	120	20	2280-2281	ΔX_{lat}	8		
	20-21	0_u	120	20	2282-2283	"	8		
	22-30	60	120	0	2284-2292	α: -4 to +9	-		
31	3-4	60	120	40	2296-2297	α: -4 to -2	-	:30	Cowl doors off
32	3-12	0_u	120	0	2298-2307	α: -4 to +12	-	1:00	Strut, strakes
	13-14	0_u	120	0	2308-2309	ΔX_{lat}	8		3", 6", 18", wing
	15-22	0_u	120	40	2310-2317	α: -4 to +10	-		fence--1/2 chord
	23-24	0_u	120	40	2318-2319	ΔX_{lat}	8		
33	4-10	0_u	120	0	2320-2326	α: -4 to +8	-	:20	Wing fence removed
	11-12	0_u	120	0	2327-2332	ΔX_{lat}	8		Wing - pylon gaps
									taped
34	3-11	60	80	40	2335-2343	α: -4 to +11	-	1:15	
	12-20	60	100	40	2344-2353	α: -4 to +9	-		
	21-28	60	120	40	2354-2361	α: -4 to +8	-		
35	3-7	60	120	40	2362-2366	α: -4 to +6	-	:45	Inboard fairing
									nacelle cut-off,
									gap covered
	8-11	60	120	0	2367-2370	α: -4 to +6	-		
	-	-	120	40	2371-	i_N: 0 to 90	0		
36	3-6	60	120	40	2374-2377	α: -4 to +4	-	:35	All strakes on
									nacelle removed
	7-12	60	120	0	2378-2383	α: -4 to +6	-		
	-	-	120	40	-	i_N: 0 to 90	0		
37	3-6	60	120	40	2386-2388	α: -4 to +4	0		Removed inboard na-
									celle fairing cover
	7-10	75	100	40	2389-2392	α: -4 to +4	-		
	11-16	0_u	160	0	2393-2398	α: -4 to + 11	-	:50	(Left-hand horiz.
									strut cracked)

TABLE IV.- Continued.
(Rotors off)

Run no.	Pt no.	i_N	V_{knots}	δ_F	Cntr	Sweep	α Trim	Run time	Remarks
39	4-5	60	120	40	2401-2402	α: -4 to 0	-	1:45	Nacelle original conf.
	6-8	75	100	40	2403-2405	α: -4 to +4	-		
	9-12	60	120	0	2406-2409	α: -4 to +4	-		
	13-17	-	120	40	2410-2419	i_N: 0 to 90	0		
40	3-6	60	120	40	2421-2425	α: -4 to +4	-	1:10	3" spoiler 24" along
	7-10	60	120	0	2426-2429	α: -4 to +4	-		blue stripe, out-
	11-15	-	120	40	2430-2440	i_N: 0 to 90	0		board nacelle
	16-20	0_u	160	0	2441-2447	α: -4 to +10	-		(angle of attack out)
	21-24	0_u	160	20	2448-2451	α: -4 to +8	-		
41	3-6	0_u	160	40	2454-2457	α: -4 to +6	-	:15	
42	3-5	60	120	40	2460-2462	α: -4 to -1	-	1:50	Emp. strut on
	6-8	60	120	0	2463-2465	α: -4 to -1	-		
	9-13	-	120	40	2466-2475	i_N: 0 to 90	0		
	14-17	75	100	40	2476-2479	α: -4 to +1	-		
43	4-8	0	120	0	2480-2484	α: -4 to 10	-	:45	Inlet fairing and
	9-12	0	120	20	2485-2488	α: -4 to 8	-		emp. strut on
	13-14	0	120	20	2489-2490	ΔX_{lat}	8		
	15	0	120	40	2491	-	-4		
	16-19	60	120	40	2492-2495	α: -4 to 4	-		
44	5-9	0	120	0	2496-2500	α: -4 to 10	-	:50	Rear set VGs
	10-13	0	120	20	2501-2504	α: -4 to 8	-		
	14-15	0	120	20	2505-2506	ΔX_{lat}	-		
	16-21	0	120	40	2507-2511	α: -4 to 8	-		
45	3-8	0	120	0	2514-2519	α: -4 to 11	-	1:25	Low drag VGs
	9-14	0	120	40	2520-2525	α: -4 to 8	-		
	15-19	30	140	40	2526-2531	α: -4 to 8	-		
	20-25	90	80	40	2532-2539	α: -4 to 12	-		
46	5-9	0	120	0	2539-2543	α: -4 to 10	-	:25	Nose boom, pylon
	10-13	0	120	40	2544-2547	α: -4 to 8	-		strake, tab flap,
									fuel line removed
47	3-7	0	120	0	2548-2552	α: -4 to 10	-	:25	Gaps taped
	8-11	0	120	40	2553-2556	α: -4 to 8	-		
48	3-5	-	80	40	2560-2562	i_N = 0 to 90	-4	1:10	Large nacelle
									strake 15 x 52"
	6-16	60	120	40	2563-2573	α = -4 to 10	-		
	17-21	60	120	40	2574-2578	ψ = -6 to +8	+6		
	22-26	0	120	40	2579-2583	ψ = -4 to +12	+6		
49	3	0	120	40	2586	-	-4	:25	Large nacelle
	4-8	60	120	40	2587-2591	α = -4 to +8	-		strake 15 x 67"
									wing fence 12"

TABLE IV.- Concluded.
(Rotors off)

Run no.	Pt no.	i_N	V_{knots}	δ_F	Cntr	Sweep	α trim	Run time	Remarks
50	3-13	60	120	40	2592-2603	α = -4 to +8	-	:30	Removed wing fence
51	3-10	0	120	0	-	α = -4 to +10	-	:30	Removed strake, VGs, strut, gaps taped
52	-	-	-	-		tunnel tare	-	-	

TABLE V.– SCALE DATA

RUN 3 TEST 525

PT	VELO / QPSF / MACH / TEMP / KNOTS	ACFSEC / PSI / RHO / RHO	INA / PFC POS / LONG STR / LAT STR / POS CE	LIFT / LIFT / LIFT/Q / CL	DRAG / DRAG / DRAG/Q / FE / CD	SIDE / SIDE / SIDE/Q / CY	PITCH / PITC / PITCH/Q / CM	YAW / YAW / YAW/Q / CN	ROLL / ROLL / ROLL/Q / CROLL
2	82.3	-32.0	90.0	-1025.	1462.	-12.	5573.	76.	344.
	21.9	0.0	50.0	-1025.	1462.	-12.	5573.	76.	344.
	0.121		50.0	-1076.	1555.	-13.	5174.	82.	366.
	90.	0.2220	50.0		64.25				
	683.1	0.9380	50.1	-0.2592	0.3771	-0.0031	0.4207	0.0208	0.0028
3	80.3	-6.0	90.0	195.	1262.	-32.	6645.	176.	432.
	20.5	0.0	50.0	195.	1262.	-32.	6645.	176.	432.
	0.118		50.0	150.	1352.	-34.	7151.	189.	464.
	93.	0.2209	50.0		52.26				
	683.5	0.9293	50.0	0.3395	0.3880	-0.0087	0.3447	0.0045	0.0037
4	80.1	-4.0	90.0	1310.	1289.	-26.	6425.	87.	425.
	20.2	0.0	50.0	1310.	1289.	-26.	6425.	87.	425.
	0.117		50.0	1411.	1387.	-28.	6915.	93.	457.
	93.	0.2209	50.0		53.46				
	683.0	0.9293	50.0	0.3588	0.3530	-0.0055	0.3349	0.0017	0.0056
5	80.0	-2.0	90.0	1905.	1230.	10.	5957.	-114.	405.
	20.1	0.0	50.0	1905.	1230.	10.	5957.	-114.	405.
	0.117		50.0	2050.	1324.	11.	6410.	-123.	546.
	93.	0.2209	50.0		51.66				
	683.0	0.9293	50.0	0.5223	0.3373	0.0027	0.3107	[illegible]	0.0040
6	80.3	0.0	90.0	2470.	1292.	23.	5561.	-175.	433.
	20.3	0.0	50.0	2470.	1292.	23.	5561.	-175.	433.
	0.119		49.9	2663.	1393.	25.	5766.	-189.	466.
	94.	0.2205	50.0		65.14				
	683.6	0.9276	50.0	0.6733	0.3522	0.0063	0.2761	-0.0015	0.0037
7	80.2	2.0	90.0	2985.	1302.	13.	4706.	144.	450.
	20.2	0.0	50.0	2985.	1302.	13.	4706.	144.	650.
	0.118		50.0	3215.	1400.	14.	5073.	155.	482.
	94.	0.2205	50.0		64.59				
	683.6	0.9276	50.1	0.8156	0.3557	0.0036	0.2407	0.0012	0.0071
8	80.2	4.0	90.0	3530.	1364.	-21.	3936.	376.	775.
	20.2	0.0	50.0	3530.	1364.	-21.	3936.	376.	775.
	0.118		50.0	3805.	1470.	-23.	4243.	405.	836.
	94.	0.2205	50.1		67.45				
	683.6	0.9276	50.0	0.9645	0.3727	-0.0057	0.2046	0.0032	0.0066
9	80.4	6.0	90.0	4415.	1522.	-17.	1408.	226.	466.
	20.3	0.0	50.0	4415.	1522.	-17.	1408.	226.	466.
	0.118		50.0	4766.	1644.	-18.	1521.	244.	527.
	95.	0.2201	50.0		75.69				
	684.2	0.9259	50.0	1.2034	0.4149	-0.0046	0.0750	0.0019	0.0041
10	80.3	12.0	90.0	4890.	1755.	47.	-1715.	-159.	406.
	20.2	0.0	50.0	4890.	1755.	47.	-1715.	-159.	405.
	0.117		49.7	5290.	1931.	51.	-1855.	-172.	442.
	96.	0.2197	50.0		66.48				
	684.8	0.9243	50.0	1.3342	0.4889	0.0129	-0.0844	-0.0014	0.0035
11	80.1	13.0	90.0	5000.	1884.	34.	-2980.	26.	225.
	20.1	0.0	50.0	5000.	1884.	34.	-2980.	26.	225.
	0.117		49.3	5409.	2036.	37.	-3230.	28.	244.
	96.	0.2197	50.0		43.73				
	684.8	0.9243	50.0	1.3743	0.5178	0.0093	-0.1502	0.0002	0.0019
12	80.2	14.0	90.0	5092.	2002.	64.	-3832.	116.	559.
	20.1	0.0	50.0	5092.	2002.	64.	-3832.	116.	559.
	0.117		49.0	5567.	2165.	69.	-4148.	126.	605.
	96.	0.2197	50.0		49.40				
	684.8	0.9243	50.0	1.3973	0.5495	0.0176	-0.2302	0.0010	0.0049
13	79.8	-4.0	90.0	165.	1260.	-434.	6689.	3954.	-1079.
	19.8	-4.0	50.0	165.	1260.	-434.	6689.	3954.	-1079.
	0.117		49.4	179.	1372.	-471.	7265.	4294.	-1172.
	98.	0.2189	50.0		63.42				
	686.0	0.9208	50.0	0.0460	0.3532	-0.1209	0.3545	0.0342	-0.0045
14	79.9	-4.0	90.0	135.	1254.	-230.	6525.	2030.	-455.
	19.9	-2.0	50.0	135.	1254.	-230.	6525.	2030.	-455.
	0.117		49.4	147.	1362.	-250.	7089.	2205.	-494.
	98.	0.2188	50.0		62.49				
	686.0	0.9204	50.0	0.0375	0.3480	-0.0638	0.3446	0.0175	-0.0034
15	80.0	-8.0	90.0	115.	1244.	191.	6959.	-1802.	1116.
	19.9	2.0	50.0	115.	1244.	191.	6959.	-1802.	1116.
	0.117		49.6	125.	1352.	208.	7561.	-1958.	1212.
	98.	0.2188	50.0		62.41				
	686.0	0.9204	50.0	0.0319	0.3448	0.0529	0.3671	-0.0155	0.0092
16	80.0	-4.0	90.0	120.	1261.	343.	6992.	-3510.	1964.
	19.9	4.0	50.1	120.	1261.	343.	6992.	-3510.	1964.
	0.117		49.6	130.	1370.	372.	7597.	[illegible]	2134.
	98.	0.2188	50.0		63.28				
	686.0	0.9204	50.0	0.0333	0.3495	0.1062	0.3697	-0.0320	0.0169
17	79.9	-8.0	90.0	105.	1301.	774.	7322.	-7435.	2737.
	19.9	8.0	50.0	105.	1301.	774.	7322.	-7435.	2737.
	0.117		49.6	114.	1413.	841.	7955.	-8074.	[illegible]
	98.	0.2188	50.1		65.35				
	686.0	0.9204	50.0	0.0291	0.3610	0.2148	0.3667	[illegible]	0.0255
18	79.8	-12.0	90.0	85.	1374.	1157.	[illegible]	[illegible]	[illegible]
	19.8	12.0	50.0	85.	1374.	1157.	[illegible]	[illegible]	[illegible]
	0.116		49.0	93.	1495.	1259.	[illegible]	[illegible]	[illegible]
	99.	0.2184	50.0		69.55				
	686.7	0.9189	50.0	0.0237	0.3831	0.3226	[illegible]	[illegible]	[illegible]
19	80.3	-16.0	90.0	-11.	1497.	[illegible]	8426.	[illegible]	5500.
	20.1	16.0	50.0	-11.	1497.	[illegible]	8426.	[illegible]	5500.
	0.117		49.4	-11.	1627.	[illegible]	[illegible]	[illegible]	5985.
	99.	0.2184	50.1		78.56				
	686.7	0.9187	50.0	-0.0028	0.4139	[illegible]	[illegible]	[illegible]	0.0471

TABLE V.— Continued.

RUN 3 TEST 525

PT	VATS QPSF MACH TEMP +SQUATS	ALPHA,C PSI RHO RHOR	INA PED POS LONG STK LAT STK POW LEV	LIFT,O LIFT LIFT/6 CL	DRAG,O DRAG DRAG/6 FE CD	SIDE,O SIDE SIDE/6 CY	PITCH,O PITCH PITCH/6 CM	YAW,O YAW YAW/6 CN	ROLL,O ROLL ROLL/6 CROLL
20	80.2 20.0 0.117 100. 687.3	-8.0 0.0 0.2180 0.9171	90.0 50.0 49.6 49.3 50.0	-80. -80. -87. -0.0221	1274. 1274. 384. 63.80 0.3527	-22. -22. -24. -0.0061	10754. 10754. 11136. 0.5670	77. 77. 64. 0.0007	533. 533. 501. 0.0045
21	80.2 20.0 0.117 100. 687.3	-8.0 0.0 0.2180 0.9171	90.0 50.0 49.1 49.7 50.0	350. 350. 362. 0.0969	1220. 1220. 1330. 61.06 0.3373	-23. -23. -25. -0.0059	2724. 2724. 2475. 0.1436	242. 242. 254. 0.0062	558. 558. 600. 0.0048
22	80.1 19.9 0.117 100. 687.3	-8.0 0.0 0.2180 0.9172	90.0 50.1 49.9 [illegible] 50.0	130. 130. 142. 0.0360	1249. 1249. 1362. 62.56 0.3462	-52. -52. -57. -0.0144	6251. 6251. 6815. 0.3247	-138. -138. -151. -0.0012	2345. 2345. 2557. 0.0202
23	81.2 20.3 0.117 100. 687.3	-8.0 0.0 0.2180 0.9171	90.0 50.0 49.4 65.0 50.0	30. 30. 33. 0.0083	1261. 1261. 1375. 63.11 0.3487	-70. -70. -76. -0.0194	6561. 6561. 7154. 0.3456	20. 20. 22. 0.0002	4330. 4330. 4721. 0.0372
24	80.2 20.0 0.117 100. 687.3	-8.0 0.0 0.2180 0.9171	90.0 71.0 50.0 50.1 50.0	155. 155. 169. 0.0429	1249. 1249. 1362. 62.51 0.3454	-163. -163. -178. -0.0451	6634. 6634. 7240. 0.3477	3280. 3280. 3583. 0.0262	-578. -578. -634. -0.0150
25	80.2 20.0 0.117 100. 687.3	-8.0 0.0 0.2180 0.9171	90.0 91.0 50.0 50.1 49.9	135. 135. 147. 0.0372	1253. 1253. 1377. 63.29 0.3466	-265. -265. -289. -0.0734	7018. 7018. 7652. 0.3597	5947. 5947. 6534. 0.0516	-1782. -1782. -1943. -0.0153

RUN 5 TEST 525

PT	VATS QPSF MACH TEMP +SQUATS	ALPHA,C PSI RHO RHOR	INA PED POS LONG STK LAT STK POW LEV	LIFT,O LIFT LIFT/6 CL	DRAG,O DRAG DRAG/6 FE CD	SIDE,O SIDE SIDE/6 CY	PITCH,O PITCH PITCH/6 CM	YAW,O YAW YAW/6 CN	ROLL,O ROLL ROLL/6 CROLL
2	60.7 11.6 0.089 93. 683.0	-12.0 0.0 0.2216 0.9324	90.0 50.0 50.0 50.0 50.0	115. 115. 123. 0.0547	943. 943. 1011. 81.17 0.4484	29. 29. 31. 0.0136	4667. 4667. 5045. 0.4623	-50. -50. -53. -0.3002	-52. -52. -56. -0.0008
3	60.9 11.7 0.089 93. 683.0	-8.0 0.0 0.2216 0.9323	90.0 50.0 50.0 50.0 50.0	750. 750. 804. 0.3544	920. 920. 987. 75.76 0.4346	12. 12. 13. 0.0057	4237. 4237. 4544. 0.3802	-204. -204. -219. -0.0030	-84. -84. -95. -0.0013
4	61.0 11.6 0.090 93. 683.0	-4.0 0.0 0.2216 0.9323	90.0 50.0 50.0 50.0 50.0	1360. 1360. 1459. 0.6467	918. 918. 985. 76.04 0.4312	0. 0. 0. 0.0026	3463. 3463. [illegible] 0.3542	-257. -257. -276. -0.0036	-306. -306. -326. -0.0045
5	60.5 11.5 0.088 94. 683.6	-2.0 0.0 0.2212 0.9307	90.0 50.0 50.0 50.0 50.0	1660. 1660. 1784. 0.7945	893. 893. 959. 77.64 0.4501	21. 21. 23. 0.0101	3737. 3737. 4015. 0.3625	-114. -114. -122. -0.0017	-138. -138. -149. -0.0021
6	60.3 11.4 0.088 94. 683.6	0.0 0.0 0.2212 0.9307	90.0 50.0 50.0 50.0 50.1	1970. 1970. 2117. 0.9458	927. 927. 975. 74.23 0.4517	14. 14. 15. 0.0068	3297. 3297. 3558. 0.3625	-147. -147. -158. -0.0022	-29. -29. -31. -0.0004
7	60.5 11.5 0.088 94. 683.6	2.0 0.0 0.2212 0.9307	90.0 50.0 50.0 50.0 50.0	2230. 2230. 2396. 1.0740	936. 936. 1006. 81.59 0.4506	5. 5. 5. 0.0024	2828. 2828. 3038. 0.2541	-335. -335. -360. -0.0054	53. 53. 57. 0.0008
8	60.4 11.5 0.088 94. 683.6	4.0 0.0 0.2212 0.9307	90.0 50.0 50.0 50.0 50.0	2515. 2515. 2702. 1.2087	973. 973. 1045. 84.64 0.4676	12. 12. 13. 0.0056	2333. 2333. 2507. 0.2134	-327. -327. -351. -0.0049	-20. -20. -21. -0.0003
9	60.3 11.4 0.088 95. 684.2	8.0 0.0 0.2208 0.9291	90.0 50.0 50.0 50.1 50.0	3035. 3035. 3267. 1.4684	1051. 1051. 1131. 91.41 0.5072	33. 33. 36. 0.0159	1315. 1315. 1415. 0.1204	-131. -131. -141. -0.0020	-25. -25. -27. -0.0004
10	60.4 11.5 0.088 95. 684.2	12.0 0.0 0.2208 0.9291	90.0 50.0 50.0 50.1 50.0	3140. 3140. 3380. 1.5177	1215. 1215. 1308. 105.47 0.5851	36. 36. 41. [illegible]	-745. -745. -802. [illegible]	83. 83. 89. [illegible]	85. 85. 91. 0.0013
11	60.5 11.5 0.088 95. 684.2	16.0 0.0 0.2208 0.9291	90.0 50.0 49.6 50.1 50.0	3205. 3205. 3450. 1.5461	1290. 1290. 1389. 112.21 0.6199	33. 33. 36. 0.0159	-1517. -1517. -1633. [illegible]	294. 294. 316. [illegible]	-404. -404. -435. [illegible]

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TABLE V.— Continued.

RUN 5 TEST 525

PT	VKTS QPSF MACH TEMP VSNDKTS	ALFS,C PSI RHO*100 RHOR	THA PED POS LONG STK LAT STK POW LEV	LIFT,U LIFT LIFT/Q CL	DRAG,U DRAG DRAG/Q FE CD	SIDE,U SIDE SIDE/Q CY	PITCH,U PITCH PITCH/Q CM	YAW,U YAW YAW/Q CN	ROLL,U ROLL ROLL/Q CROLL
12	60.4 11.5 0.088 95. 684.2	14.[illegible] 3.[illegible] 0.2204 0.924[illegible]	[illegible] 50.[illegible] 49.6 [illegible] [illegible]	3[illegible] 3[illegible] 3364. 1.5070	1382. 1382. 1466. 120.47 0.5656	17. 17. 18. -0.1042	-2562. -2562. -2705. -0.2571	[illegible] [illegible] [illegible] [illegible]	-759. -759. [illegible] [illegible]
13	60.6 11.5 0.089 95. 680.2	[illegible] [illegible] 0.2208 [illegible]	[illegible] 50.2 49.4 5[illegible] 51.0	735. 735. 741. 0.3517	9[illegible] 9[illegible] 960. 74.82 0.4355	-235. -235. -231. -0.1115	4624. 4624. 4477. 0.4211	2315. 2315. 2491. 0.[illegible]344	-1040. -10[illegible] -1119. [illegible]
14	60.4 11.5 0.088 96. 684.8	[illegible] [illegible] 0.2204 0.9274	90.0 50.2 44.4 50.1 50.[illegible]	740. 740. 744. 0.3562	564. 564. 464. 77.43 0.[illegible]375	-127. -127. -137. -0.0012	4625. 4625. 4487. 0.4215	1164. 1164. 1281. 0.0175	-541. -541. -422. -0.0154
15	60.4 11.4 0.088 96. 684.8	-9.6 2.0 0.2204 0.9274	90.0 50.[illegible] 50.0 50.2 50.0	735. 735. 743. 0.3547	84[illegible] 84[illegible] 96[illegible] 77.75 0.4255	123. 123. 133. 0.0594	4844. 4844. 5224. 0.4454	-1324. -1324. -1423. -0.0172	445. 445. 46[illegible] [illegible]
16	60.4 11.5 0.088 96. 684.8	-5.0 -3.0 0.2204 0.9274	90.0 50.2 50.0 50.1 50.0	721. 720. 776. 0.3466	90[illegible] 90[illegible] 970. 78.43 0.4358	247. 247. 266. 0.1196	4764. 4764. 5142. 0.4376	-2521. -2521. -2719. -0.0377	826. 826. 893. 0.0124
17	60.4 11.4 0.088 96. 684.8	[illegible] 6.0 0.2204 0.9274	90.0 50.1 50.0 50.[illegible] 50.0	686. 686. 735. 0.3292	917. 917. 984. 80.16 0.4426	447. 447. 482. 0.2157	443[illegible] 443[illegible] 5525. 0.4555	-4556. -4556. -4915. [illegible]	1227. 1227. 1315. [illegible]
18	60.0 11.3 0.088 97. 685.4	-8.0 12.0 0.2200 0.9258	90.0 50.2 50.0 50.[illegible] 50.0	615. 615. 664. 0.3013	955. 955. 1026. 84.24 0.4654	632. 632. 683. 0.3096	4576. 4576. 4943. 0.4266	-6291. -6291. -6795. -0.0956	2188. 2188. 2361. 0.0330
19	60.3 11.4 0.088 97. 685.4	-8.0 16.0 0.2200 0.9257	90.0 50.[illegible] 50.0 50.2 50.0	545. 545. 589. 0.2641	1038. 1038. 1121. 91.00 0.5031	723. 723. 781. 0.3504	4675. 4675. 5050. 0.4312	-6944. -6944. -7500. -0.1047	2236. 2236. 2415. 0.0340
20	60.7 11.5 0.089 97. 685.4	-8.0 0.0 0.2200 0.9257	90.0 50.[illegible] 50.0 65.2 50.0	730. 730. 789. 0.3495	898. 898. 965. 77.81 0.4298	-15. -15. -16. -0.0072	4431. 4431. 4787. 0.6035	-21[illegible] -21[illegible] [illegible] [illegible]	-15. -15. [illegible] [illegible]
21	60.7 11.5 0.089 97. 685.4	-6.0 0.0 0.2200 0.9257	90.0 50.[illegible] 50.0 65.[illegible] 50.0	595. 595. 647. 0.306[illegible]	886. 886. 957. 76.76 0.4290	-42. -42. -45. -0.0201	4357. 4357. 4707. [illegible]	-411. -411. -444. [illegible]	[illegible] 1303. 1404. [illegible]

RUN 6 TEST 525

PT	VKTS QPSF MACH TEMP VSNDKTS	ALFS,C PSI RHO*100 RHOR	THA PED POS LONG STK LAT STK POW LEV	LIFT,U LIFT LIFT/Q CL	DRAG,U DRAG DRAG/Q FE CD	SIDE,U SIDE SIDE/Q CY	PITCH,U PITCH PITCH/Q CM	YAW,U YAW YAW/Q CN	ROLL,U ROLL ROLL/Q CROLL
5	159.9 79.9 0.237 87. 679.2	[illegible] 0.0 0.2193 0.9225	0.0 50.[illegible] 49.4 49.4 49.[illegible]	1035. 1035. 1122. 0.0736	1954. 1954. 2116. 24.47 0.1352	65. 65. 66. 0.0044	21562. 21562. [illegible] 0.2644	3584. 3584. [illegible] [illegible]	2915. 2915. [illegible] [illegible]
6	160.2 79.8 0.237 89. 680.5	[illegible] 0.0 0.2165 0.919[illegible]	0.0 50.1 49.4 49.9 [illegible]	3695. 3695. 4020. 0.25[illegible]	1787. 1787. 1944. 22.30 0.1237	-41. -41. -45. [illegible]	17935. 17935. [illegible] 0.2362	3975. 3975. 4325. [illegible]	5032. 5032. 5471. [illegible]
7	160.2 79.7 0.237 [illegible] [illegible]	[illegible] 0.0 0.21[illegible] 0.9175	[illegible] 50.1 49.4 [illegible] [illegible]	6135. 6135. 6667. 0.4255	[illegible] [illegible] [illegible] 21.03 [illegible]	[illegible] [illegible] [illegible] [illegible]	15139. 15139. [illegible] [illegible]	[illegible] [illegible] [illegible] [illegible]	[illegible] [illegible] [illegible] [illegible]
8	160.6 79.9 0.237 91. 681.7	[illegible] [illegible] [illegible] 0.9158	[illegible] 50.[illegible] 49.4 [illegible] 49.4	8860. 8860. 9675. 0.6125	[illegible] [illegible] [illegible] [illegible] [illegible]	[illegible]	[illegible]	[illegible]	[illegible]
9	160.8 80.0 0.237 92. 682.3	[illegible] 0.0 0.2175 0.914[illegible]	[illegible] 50.0 49.4 49.4 49.7	11535. 11535. 12519. 0.7964	1833. 1833. [illegible] 22.41 [illegible]	[illegible]	[illegible]	[illegible]	[illegible]

TABLE V.— Continued

RUN 6 TEST 525

PT	[illegible] QPSF [illegible] TEMP VSNDKTS	ALPHA PSI MACH [illegible]	[illegible] PET POS CONV STA LAT STA POP LEV	LIFT LIFT LIFT/B CL	DRAG DRAG DRAG/B RE CD	SIDE SIDE SIDE/B [illegible]	PITCH PITCH PITCH/B [illegible]	YAW YAW YAW/B CN	ROLL ROLL ROLL/B CROLL
10	161.2	[illegible]	[illegible]	10025.	2225.	-32.	[illegible]	1463.	5534.
	80.3	0.0	50.1	10025.	2225.	-32.	[illegible]	1463.	5534.
	0.236		49.[illegible]	11505.	2459.	-35.	[illegible]	2066.	6066.
	93.	0.2164	50.[illegible]		27.78				
	683.0	0.9123	49.8	1.1030	0.1531	-0.0022	-0.034[illegible]	[illegible]	[illegible]
11	121.5	-4.0	0.0	585.	1096.	4.	[illegible]	1536.	2200.
	46.3	0.0	50.2	585.	1096.	4.	[illegible]	1536.	2200.
	0.179		49.1	632.	1187.	10.	[illegible]	1600.	2379.
	92.	0.2199	50.0		23.70				
	682.3	0.9251	49.7	0.0698	0.1309	[illegible]	[illegible]	[illegible]	0.0062
12	121.4	-2.0	0.0	1970.	1024.	-8.	9195.	1679.	2402.
	46.5	0.0	50.1	1970.	1024.	-8.	9195.	1679.	2402.
	0.179		49.0	2133.	1107.	-9.	[illegible]	1815.	2597.
	92.	0.2199	50.0		22.65				
	682.3	0.9250	49.7	0.2391	0.1277	-0.0010	0.2215	[illegible]	0.0069
13	122.0	0.0	0.0	3475.	976.	-45.	7001.	2027.	2650.
	46.7	0.0	50.2	3475.	976.	-45.	7001.	2027.	2650.
	0.180		49.1	3750.	1051.	-49.	[illegible]	2187.	2901.
	91.	0.2203	50.0		22.07				
	681.7	0.9266	49.7	0.4113	0.1153	-0.0053	0.1777	0.0075	0.0099
14	122.0	2.0	0.0	4935.	966.	-28.	4932.	1566.	4155.
	46.7	0.0	50.1	4935.	966.	-28.	4932.	1566.	4155.
	0.180		49.1	5326.	1045.	-3[illegible]	5323.	1690.	4484.
	91.	0.2203	50.0		20.72				
	681.7	0.9266	49.7	0.5836	0.1145	-0.0033	0.1110	0.0058	0.0153
15	121.9	4.0	0.0	6245.	1016.	-30.	371[illegible]	1466.	3651.
	46.6	0.0	50.1	6245.	1016.	-30.	371[illegible]	1466.	3651.
	0.179		49.1	6739.	1099.	-32.	[illegible]	1603.	3940.
	91.	0.2203	50.0		21.85				
	681.7	0.9267	49.6	0.7406	0.1207	-0.0036	[illegible]	0.0055	0.0135
16	121.6	6.0	0.0	8750.	1226.	-24.	-7648.	1076.	3691.
	46.3	0.0	50.2	8750.	1226.	-24.	-7648.	1076.	3691.
	0.179		49.0	9441.	1323.	-26.	-8254.	1161.	4190.
	91.	0.2203	50.1		26.50				
	681.7	0.9268	49.7	1.0450	0.1464	-0.0029	-0.0902	0.0040	0.0149
17	120.6	12.0	0.0	10155.	1597.	-17.	-4914[illegible]	160.	2376.
	45.8	0.0	50.1	10155.	1597.	-17.	-4914[illegible]	160.	2376.
	0.176		48.9	10956.	1723.	-16.	[illegible]	172.	2565.
	91.	0.2203	50.0		34.65				
	681.7	0.9269	49.7	1.2240	0.1925	-0.0020	-0.2110	0.0006	0.0069
18	161.2	-4.0	0.0	7950.	2461.	-171.	22970.	2443.	1901.
	80.0	0.0	50.1	7950.	2461.	-171.	22970.	2443.	1901.
	0.237		48.9	8745.	2703.	-159.	25260.	2686.	2157.
	95.	0.2161	50.1		30.75				
	684.2	0.9091	49.7	0.5989	0.1699	-0.0116	0.3016	0.0052	0.0042
19	161.2	-2.0	0.0	10795.	2460.	-163.	21285.	1510.	1237.
	80.0	0.0	50.1	10795.	2460.	-163.	21285.	1510.	1237.
	0.237		49.0	11874.	2706.	-174.	23412.	1661.	1338.
	95.	0.2161	50.1		30.76				
	684.2	0.9091	49.6	0.7457	0.1699	-0.0113	0.2798	0.0032	0.0026
20	161.1	0.0	0.0	13555.	2526.	-146.	18201.	2299.	3312.
	79.8	0.0	50.1	13555.	2526.	-146.	18201.	2299.	3312.
	0.237		49.0	14930.	2783.	-151.	20055.	2533.	3649.
	96.	0.2157	50.0		31.67				
	684.8	0.9075	49.6	0.9388	0.1750	-0.0101	0.2349	0.0049	0.0071
21	161.0	2.0	0.0	16445.	2736.	-183.	14232.	2215.	1690.
	79.9	0.0	50.1	16445.	2736.	-183.	14232.	2215.	1690.
	0.237		48.9	18153.	2487.	+262.	5710.	2440.	[illegible]
	97.	0.2155	50.[illegible]		33.86				
	685.0	0.9059	49.6	1.1389	0.1871	-0.0127	0.1842	0.0048	0.0036
22	161.6	4.0	0.0	19205.	2940.	-167.	10900.	1452.	2291.
	79.9	0.0	50.1	19205.	2940.	-167.	10900.	1452.	2291.
	0.237		49.0	21240.	3252.	-165.	12[illegible]	[illegible]	253[illegible]
	98.	0.2149	50.0		36.78				
	686.0	0.9039	49.6	1.3274	0.2032	-0.0115	[illegible]	0.0042	[illegible]
23	162.0	8.0	0.0	23845.	3736.	+266.	-27[illegible]	[illegible]	[illegible]
	80.1	0.0	50.1	23845.	3736.	+256.	-27[illegible]	[illegible]	[illegible]
	0.237		49.[illegible]	26420.	4141.	-245.	[illegible]	[illegible]	[illegible]
	99.	0.2145	50.1		46.65				
	686.7	0.9023	49.5	1.6442	0.2576	-0.0183	-0.0363	0.0045	[illegible]
24	104.4	-4.0	0.0	3625.	1074.	-52.	[illegible]	1213.	1165.
	37.9	0.0	50.1	3625.	1074.	-52.	[illegible]	1213.	1165.
	0.162		49.1	3903.	1156.	-55.	[illegible]	1306.	1276.
	91.	0.2202	50.1		28.36				
	681.7	0.9266	49.5	0.5277	0.1564	[illegible]	0.2829	[illegible]	[illegible]
25	104.4	-2.0	0.0	4920.	1073.	-71.	4535.	1657.	1145.
	37.9	0.0	50.1	4920.	1073.	-71.	4535.	1657.	1145.
	0.162		49.1	5297.	1155.	-76.	[illegible]	1136.	1255.
	91.	0.2204	50.0		28.2[illegible]				
	681.7	0.9268	49.5	0.7161	0.1562	[illegible]	[illegible]	[illegible]	0.0052
26	104.4	0.0	[illegible]	6135.	1146.	-67.	[illegible]	785.	1116.
	37.9	0.0	50.1	6135.	1146.	-67.	[illegible]	785.	1116.
	0.162		49.1	6605.	1234.	-72.	[illegible]	845.	1201.
	91.	0.2206	50.[illegible]		30.22				
	681.7	0.9269	49.5	0.8937	0.1669	[illegible]	0.2[illegible]	[illegible]	0.0051
27	104.8	2.0	0.0	7476.	1209.	-135.	5643.	1552.	894.
	37.9	0.0	50.1	7476.	1209.	-135.	5643.	1552.	[illegible]
	0.162		49.1	8042.	1302.	-113.	[illegible]	1430.	[illegible]
	91.	0.2204	50.1		31.90				
	681.7	0.9269	49.5	1.0906	0.1761	[illegible]	[illegible]	[illegible]	[illegible]

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TABLE V.— Continued.

RUN 6 TEST 525

PT	VEAS / QPSF / MACH / TEMP / VSNDKTS	ALFS,C / PSI / / RHO,LOG / RHOR	INA / PED POS / LONG STA / LAT STA / POS LEV	LIFT,U / LIFT / LIFT/8 / / CL	DRAG,U / DRAG / DRAG/8 / FE / CD	SIDE,U / SIDE / SIDE/8 / / CY	PITCH,U / PITCH / PITCH/8 / / CM	YAW,U / YAW / YAW/8 / / CN	ROLL,U / ROLL / ROLL/8 / / CROLL
28	104.9	4.0	0.0	8660.	1341.	+76.	5825.	1020.	1194.
	38.2	0.0	50.1	8660.	1341.	+76.	5825.	1020.	1194.
	0.162		49.1	9302.	1444.	+82.	6116.	1094.	1176.
	91.	0.2208	50.1		35.35				
	681.7	0.9288	49.5	0.2571	0.1951	+0.0111	0.1459	0.0046	0.0049
29	110.1	8.0	0.0	10770.	1668.	-119.	-1423.	857.	874.
	38.1	0.0	50.1	10770.	1668.	-119.	-1423.	857.	874.
	0.162		49.1	11590.	1796.	-128.	[illegible]	933.	941.
	91.	0.2208	50.1		45.74				
	681.7	0.9288	49.5	0.5622	0.2919	-0.0173	-0.1531	[illegible]	[illegible]

RUN 7 TEST 525

PT	VEAS / QPSF / MACH / TEMP / VSNDKTS	ALFS,C / PSI / / RHO,LOG / RHOR	INA / PED POS / LONG STA / LAT STA / POS LEV	LIFT,U / LIFT / LIFT/8 / / CL	DRAG,U / DRAG / DRAG/8 / FE / CD	SIDE,U / SIDE / SIDE/8 / / CY	PITCH,U / PITCH / PITCH/8 / / CM	YAW,U / YAW / YAW/8 / / CN	ROLL,U / ROLL / ROLL/8 / / CROLL
4	93.7	0.0	0.0	2295.	637.	+21.	4799.	1403.	1426.
	32.3	0.0	50.1	2295.	637.	-21.	4799.	1403.	1426.
	0.144		50.1	2376.	670.	+21.	4967.	1455.	1494.
	71.	0.2296	50.0		19.46				
	669.2	0.9680	49.8	0.3499	0.1099	+0.0034	0.1551	0.0074	0.0132
5	131.4	0.0	0.0	4090.	1139.	-32.	6671.	2183.	2035.
	55.5	0.0	50.1	4090.	1139.	-32.	6671.	2183.	2035.
	0.196		50.1	4306.	1200.	-34.	9134.	2244.	2495.
	76.	0.2256	50.1		20.54				
	672.6	0.9493	49.7	0.6075	0.1135	-0.0032	0.1646	0.0068	0.0088
6	158.7	0.0	0.0	6015.	1718.	-76.	13547.	4455.	5717.
	79.4	0.0	50.3	6015.	1718.	-76.	13547.	4455.	5717.
	0.236		50.3	6464.	1846.	-82.	14559.	4787.	6144.
	52.	0.2212	50.1		21.65				
	676.1	0.9305	49.4	0.4191	0.1196	-0.0053	0.1795	0.0046	0.0124
7	176.7	0.0	0.0	7570.	2154.	-21.	18086.	4579.	5716.
	95.6	0.0	57.0	7570.	2154.	-21.	18086.	4579.	5716.
	0.263		50.4	8365.	2380.	-23.	19985.	5060.	6317.
	94.	0.2151	50.2		22.52				
	683.5	0.9050	49.4	0.4379	0.1294	-0.0012	0.1988	0.0052	0.0103

RUN 8 TEST 525

PT	VEAS / QPSF / MACH / TEMP / VSNDKTS	ALFS,C / PSI / / RHO,LOG / RHOR	INA / PED POS / LONG STA / LAT STA / POS LEV	LIFT,U / LIFT / LIFT/8 / / CL	DRAG,U / DRAG / DRAG/8 / FE / CD	SIDE,U / SIDE / SIDE/8 / / CY	PITCH,U / PITCH / PITCH/8 / / CM	YAW,U / YAW / YAW/8 / / CN	ROLL,U / ROLL / ROLL/8 / / CROLL
3	79.7	0.0	0.0	2375.	1284.	27.	5638.	443.	701.
	20.5	0.0	50.1	2375.	1284.	27.	5638.	443.	701.
	0.118		50.0	2491.	1347.	28.	5915.	467.	735.
	80.	0.2260	50.0		62.65				
	674.9	0.9535	50.1	0.6805	0.3463	0.0073	0.2443	[illegible]	[illegible]
4	117.2	0.0	0.0	4950.	2867.	24.	13242.	427.	1273.
	44.3	0.0	50.0	4950.	2867.	24.	13242.	427.	1273.
	0.175		50.0	5244.	3038.	25.	14029.	452.	1349.
	81.	0.2243	50.0		64.60				
	675.5	0.9439	50.1	0.6170	0.3574	0.0030	0.3141	[illegible]	0.0049
5	111.2	0.0	0.0	6420.	1190.	-21.	8098.	721.	756.
	39.5	0.0	50.0	6420.	1190.	-21.	8098.	721.	756.
	0.165		50.0	6824.	1266.	-22.	8613.	766.	755.
	84.	0.2235	50.0		30.22				
	677.4	0.9403	50.1	0.9208	0.1676	-0.0029	0.2162	0.0031	0.0032
6	170.2	0.0	0.0	15357.	2687.	-110.	20591.	1778.	2013.
	89.8	0.0	50.0	15357.	2687.	-110.	20591.	1778.	2013.
	0.252		50.0	16778.	3158.	-120.	22269.	1944.	2200.
	89.	0.2175	50.0		32.10				
	680.5	0.9149	50.0	0.9387	0.1777	-0.0048	0.2350	[illegible]	[illegible]
7	160.2	0.0	0.0	7217.	2027.	14.	16745.	3149.	4240.
	90.3	0.0	50.0	7217.	2027.	14.	16745.	3149.	4240.
	0.233		50.0	7925.	2226.	15.	18400.	3462.	4428.
	92.	0.2162	50.0		22.45				
	682.3	0.9097	50.0	0.7413	0.2241	[illegible]	0.1950	[illegible]	[illegible]
8	141.8	0.0	0.0	7110.	2151.	-52.	11346.	1154.	1576.
	66.5	0.0	50.0	7110.	2151.	-52.	11346.	1154.	1576.
	0.214		50.0	7685.	2351.	-57.	12254.	1252.	1706.
	91.	0.2200	50.0		40.41				
	681.7	0.9254	50.0	0.8444	0.2592	[illegible]	0.2564	0.0043	0.0058
9	158.6	0.0	0.0	[illegible]	4380.	-115.	25076.	2418.	2106.
	80.6	0.0	50.0	[illegible]	4380.	-115.	25076.	2418.	2106.
	0.247		49.4	15552.	4602.	-120.	25016.	2540.	2338.
	94.	0.2141	50.0		51.55				
	686.0	0.9004	50.3	0.4913	0.2753	[illegible]	[illegible]	0.0058	0.0042

TABLE V.— Continued.

RUN 9 TEST 525

PT	VEL [illegible] / DPSF / MACH / TEMP / RSQRTS	ALPHA / PSI / ANCHOR [illegible] / RHOR	[illegible] / PED POS / LONG STK / LAT STK / POW LEV	LIFT / LIFT / LIFT/Q / CL	DRAG / DRAG / DRAG/Q / L/D / CD	SIDE / SIDE / SIDE/Q / CY	PITCH / PITCH / PITCH/Q / CM	YAW / YAW / YAW/Q / CN	ROLL / ROLL / ROLL/Q / CROLL
3	[illegible]	-4.0	0.0	200.	501.	[illegible]	[illegible]	[illegible]	1313.
	21.2	0.0	50.[illegible]	200.	501.	[illegible]	[illegible]	908.	1313.
	0.120		50.2	202.	506.	6.	5310.	[illegible]	1326.
	61.	0.2355	49.3		23.66				
	662.9	0.9908	49.9	0.0522	0.1308	0.00[illegible]5	0.2468	[illegible]	[illegible]
4	79.9	-2.0	0.0	885.	474.	3.	4561.	443.	1306.
	21.0	0.0	50.1	885.	474.	3.	4561.	443.	1306.
	0.121		50.1	895.	479.	3.	4613.	454.	1323.
	62.	0.2356	49.6		22.18				
	663.5	0.9888	50.0	0.2289	0.1225	0.0008	0.2243	[illegible]	[illegible]
5	80.0	0.0	0.0	1545.	455.	3.	3705.	[illegible]	1934.
	21.4	0.0	49.3	1585.	455.	3.	3705.	[illegible]	1934.
	0.121		50.1	1563.	460.	3.	[illegible]	1052.	1956.
	62.	0.2356	49.7		21.22				
	663.5	0.9888	49.9	0.3980	0.1172	0.0008	[illegible]	[illegible]	0.0155
6	80.1	2.0	0.0	2255.	455.	-4.	2556.	[illegible]	[illegible]
	21.4	0.0	50.5	2255.	455.	-4.	2556.	[illegible]	[illegible]
	0.121		50.2	2265.	465.	-4.	2590.	[illegible]	[illegible]
	63.	0.2346	49.6		21.41				
	664.2	0.9869	50.0	0.5878	0.1189	-0.0010	0.1255	[illegible]	0.0143
7	80.0	4.0	0.0	2920.	484.	-4.	1186.	742.	1905.
	21.4	0.0	49.1	2920.	484.	-4.	1186.	742.	1905.
	0.121		50.0	2950.	490.	-4.	1200.	752.	1929.
	63.	0.2347	49.6		22.62				
	664.2	0.9872	49.9	0.7539	0.1250	-0.0010	0.0534	0.0060	0.0153
8	80.0	6.0	0.0	3630.	524.	-16.	623.	731.	2220.
	21.4	0.0	49.9	3630.	524.	-15.	623.	731.	2220.
	0.121		50.2	3677.	531.	-16.	631.	740.	2229.
	63.	0.2347	49.6		24.49				
	664.2	0.9872	49.9	0.9372	0.1353	-0.0046	[illegible]	[illegible]	0.0177
9	80.0	8.0	0.0	4270.	583.	-4.	-172.	719.	2444.
	21.3	0.0	49.6	4270.	583.	-4.	-172.	719.	2444.
	0.121		50.1	4333.	592.	-4.	-183.	729.	2480.
	64.	0.2342	49.5		27.34				
	664.8	0.9854	50.2	1.0762	0.1510	-0.0010	[illegible]	0.0056	0.0197
10	80.0	10.0	0.0	4890.	653.	7.	-1378.	327.	1755.
	21.4	0.0	49.8	4890.	653.	7.	-1378.	327.	1755.
	0.121		50.1	4963.	663.	7.	-1396.	332.	1781.
	64.	0.2342	49.7		30.59				
	664.8	0.9853	50.0	1.2659	0.1690	0.0018	-0.0877	0.0026	0.0141
11	80.0	12.0	0.0	5480.	733.	12.	-2196.	227.	1527.
	21.3	0.0	50.0	5480.	733.	12.	-2196.	227.	1527.
	0.121		50.1	5561.	744.	12.	-2227.	231.	1550.
	64.	0.2342	49.7		34.37				
	664.8	0.9854	50.0	1.4197	0.1899	0.0031	-0.1082	[illegible]	0.0123
12	80.0	14.0	0.0	6145.	846.	2.	-3881.	210.	1799.
	21.3	0.0	51.1	6145.	846.	2.	-3881.	210.	1799.
	0.121		50.1	6236.	859.	2.	-3939.	213.	1825.
	64.	0.2342	49.7		39.67				
	664.8	0.9854	49.9	1.5919	0.2192	0.0005	[illegible]	0.0017	0.0145
13	80.0	15.0	0.0	6455.	903.	-2.	-4518.	169.	1231.
	21.3	0.0	50.2	6455.	903.	-2.	-4518.	169.	1231.
	0.121		50.2	6551.	916.	-2.	-4586.	172.	1249.
	64.	0.2342	49.6		42.34				
	664.8	0.9854	49.9	1.6723	0.2339	-0.0005	-0.2228	0.0014	0.0099
14	80.1	16.0	0.0	6370.	1074.	-54.	-5104.	-462.	-1933.
	21.3	0.0	53.7	6370.	1074.	-54.	-5108.	-462.	-1933.
	0.121		50.0	6477.	1092.	-55.	-5254.	-469.	-1965.
	65.	0.2338	49.7		50.30				
	665.4	0.9835	49.9	1.6484	0.2779	-0.0140	-0.2545	-0.0037	-0.0156
15	79.9	-4.0	0.0	2265.	664.	-37.	6324.	594.	656.
	21.3	0.0	50.9	2265.	664.	-37.	6328.	594.	656.
	0.120		50.1	2303.	675.	-39.	6434.	604.	669.
	65.	0.2338	50.0		31.26				
	665.4	0.9835	50.2	0.5888	0.1726	-0.0096	0.3130	0.0048	0.0053
16	80.0	-2.0	0.0	3025.	675.	-30.	5534.	404.	799.
	21.3	0.0	50.1	3025.	675.	-30.	5534.	404.	799.
	0.121		50.1	3076.	686.	-31.	5633.	415.	813.
	65.	0.2338	49.6		31.65				
	665.4	0.9835	50.0	0.7837	0.1749	-0.0078	0.2731	0.0033	0.0064
17	80.1	0.0	0.0	3680.	706.	-24.	4526.	411.	504.
	21.4	0.0	48.8	3680.	706.	-24.	4526.	411.	504.
	0.121		50.3	3742.	718.	-24.	4602.	418.	512.
	65.	0.2338	49.8		32.94				
	665.4	0.9835	49.9	0.9512	0.1820	-0.0062	0.2226	0.0033	0.0040
18	80.1	2.0	0.0	4360.	742.	-41.	3755.	658.	466.
	21.4	0.0	50.5	4360.	742.	-41.	3755.	658.	466.
	0.121		50.2	4435.	754.	-42.	[illegible]	618.	[illegible]
	65.	0.2335	49.7		34.75				
	665.4	0.9835	49.9	1.1282	0.1920	-0.0106	[illegible]	0.0049	0.0024
19	80.2	4.0	0.0	5060.	822.	-43.	1894.	[illegible]	544.
	21.4	0.0	49.9	5060.	822.	-43.	1895.	[illegible]	544.
	0.121		50.2	5155.	837.	-44.	1934.	[illegible]	[illegible]
	66.	0.2333	49.6		35.46				
	666.1	0.9816	49.9	1.3079	0.2125	-0.0111	[illegible]	[illegible]	0.0048
20	80.1	6.0	0.0	5675.	891.	-27.	0.	378.	700.
	21.3	0.0	51.7	5675.	891.	-27.	0.	378.	700.
	0.121		[illegible]	5781.	908.	-28.	0.	385.	713.
	66.	0.2333	49.6		[illegible]				
	666.1	0.9816	49.8	1.4702	0.2309	-0.0070	[illegible]	[illegible]	0.0056

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TABLE V.— Continued.

RUN [illegible] TEST 525

PT	[illegible] QPSF [illegible] TEMP VSNDRTS	[illegible] PSI [illegible]	[illegible] [illegible] [illegible] [illegible]	[illegible] LIFT [illegible] CL	[illegible] CD	[illegible] CY	[illegible] CM	[illegible] CN	[illegible] CROLL
21	60.2	[illegible]	[illegible]	6450.	[illegible]	-12.	[illegible]	346.	325.
	21.4	0.0	50.5	6350.	971.	-12.	79.	346.	325.
	0.12.		50.2	6571.	995.	-12.	81.	353.	329.
	66.	0.2333	49.8		45.76				
	666.1	0.9815	49.4	1.0691	0.2528	-0.0031	0.0039	0.0028	0.0026
22	60.2	[illegible]	[illegible]	7070.	[illegible]	-6.	-1145.	316.	335.
	21.4	0.0	50.4	7070.	[illegible]	-6.	-1145.	316.	335.
	0.121		50.2	7203.	1104.	-6.	-1165.	321.	341.
	60.	0.2333	49.9		50.37				
	666.1	0.9815	49.9	1.8295	0.2805	-0.0021	-0.1504	0.0025	0.0027
23	60.2	12.0	[illegible]	7660.	1203.	[illegible]	-2307.	494.	51.
	21.8	0.0	50.4	7660.	1203.	[illegible]	-2307.	494.	50.
	0.121		50.	7196.	1222.	-11.	-2355.	514.	51.
	67.	0.2329	49.6		56.55				
	666.7	0.9797	49.9	1.9770	0.3113	-0.0044	-0.3138	0.0041	0.0032
24	60.2	15.0	[illegible]	7535.	1271.	-12.	-2561.	-22.	-44.
	21.3	0.0	51.2	7535.	1271.	-12.	-2561.	-22.	-44.
	0.12.		50.1	6944.	1247.	-12.	-2635.	-2.	-501.
	67.	0.2329	50.2		54.40				
	666.7	0.9797	49.9	2.0557	0.3293	-0.0031	-0.1273	[illegible]	[illegible]
25	60.1	14.0	[illegible]	5075.	1350.	-15.	-2676.	311.	-1006.
	21.3	0.0	50.2	5075.	1350.	-15.	-2676.	311.	-1006.
	0.128		50.1	5242.	1376.	-15.	-2935.	319.	-1021.
	67.	0.2324	49.7		43.37				
	666.7	0.9797	49.9	2.0945	0.3501	-0.0039	-0.1421	0.0024	-0.0041
26	60.2	0.0	0.0	3705.	693.	-23.	4054.	484.	561.
	21.5	0.0	51.3	3705.	693.	-23.	4054.	485.	561.
	0.121		49.2	3762.	707.	-23.	4460.	494.	572.
	67.	0.2329	49.7		32.40				
	666.7	0.9797	49.8	0.9587	0.1793	-0.0060	0.2393	0.0039	0.0045
27	60.2	0.0	30.0	3495.	1030.	-51.	5407.	654.	874.
	21.4	0.0	51.3	3495.	1030.	-51.	5407.	654.	874.
	0.121		49.3	3567.	1051.	-52.	5514.	651.	897.
	67.	0.2329	49.9		48.24				
	666.7	0.9797	49.9	0.9044	0.2665	-0.0132	0.2663	0.0051	0.0071
28	60.2	0.0	45.0	3410.	1136.	13.	6347.	578.	544.
	21.3	0.0	52.3	3410.	1134.	13.	6347.	578.	544.
	0.120		49.2	3467.	1164.	13.	6516.	591.	547.
	68.	0.2324	49.5		33.48				
	667.5	0.9779	49.8	0.8854	0.2955	0.0034	0.4124	0.0047	0.0047
29	60.3	0.0	55.0	3245.	1344.	39.	6653.	330.	534.
	21.4	0.0	51.3	3245.	1393.	39.	6663.	330.	533.
	0.121		49.3	3316.	1374.	40.	6814.	336.	546.
	68.	0.2324	49.7		62.95				
	667.3	0.9779	49.9	0.8397	0.3478	0.0101	0.3281	0.0027	0.0043
30	60.3	0.0	60.0	3160.	1399.	29.	6772.	141.	559.
	21.4	0.0	51.2	3160.	1399.	29.	6772.	141.	559.
	0.12.		49.3	3231.	1431.	30.	6926.	144.	572.
	68.	0.2324	49.6		65.53				
	667.5	0.9779	49.8	0.8177	0.3620	0.0075	0.3335	0.0011	0.0045
31	60.3	0.0	75.0	2920.	1445.	15.	6852.	445.	571.
	21.3	0.0	50.4	2920.	1445.	15.	6852.	445.	571.
	0.121		49.3	2956.	1470.	15.	7007.	455.	584.
	68.	0.2324	49.4		67.70				
	667.5	0.9779	49.8	0.7565	0.3743	0.0039	0.3374	0.0036	0.0046
32	108.5	4.0	[illegible]	4225.	1196.	-67.	10852.	1146.	251.
	38.3	0.0	50.0	4225.	1196.	-67.	10852.	1146.	251.
	0.162		49.4	4387.	1242.	-70.	11264.	1190.	262.
	73.	0.2269	49.4		51.14				
	670.5	0.9630	50.0	0.6087	0.1723	-0.0097	0.2975	0.0057	0.0013
33	108.3	2.0	0.0	5450.	1224.	-61.	10196.	916.	411.
	38.3	0.0	50.2	5450.	1224.	-61.	10196.	914.	411.
	0.162		49.4	5654.	1271.	-64.	10555.	951.	426.
	73.	0.2269	49.6		32.06				
	670.5	0.9631	49.7	0.7871	0.1768	-0.0117	0.2603	0.0041	0.0041
34	108.5	0.0	[illegible]	6770.	1270.	-61.	6733.	1094.	754.
	38.3	0.0	50.7	6770.	1270.	-61.	6733.	1094.	754.
	0.162		49.6	7030.	1319.	-64.	9064.	1131.	765.
	73.	0.2269	49.7		33.72				
	670.5	0.9630	49.7	0.9754	0.1830	-0.0117	0.2394	0.0044	0.0033
35	108.5	2.0	0.0	7950.	1371.	-106.	5888.	983.	1101.
	38.3	0.0	50.4	7950.	1371.	-106.	5868.	983.	1101.
	0.162		49.5	8208.	1426.	-110.	6125.	1023.	1224.
	76.	0.2246	49.7		35.77				
	671.1	0.9616	49.7	1.1301	0.1976	-0.0153	[illegible]	0.0044	0.0052
36	108.4	4.0	[illegible]	9210.	1477.	-54.	4395.	807.	1047.
	38.2	0.0	50.3	9210.	1477.	-54.	4395.	807.	1047.
	0.162		49.5	9576.	1536.	[illegible]	4571.	832.	1090.
	74.	0.2266	49.8		34.63				
	671.1	0.9610	49.7	1.3310	0.2135	-0.0075	0.1204	0.0027	0.0049
37	108.4	0.0	[illegible]	10555.	1625.	-52.	1504.	719.	1594.
	38.4	0.0	52.0	10555.	1625.	-52.	1504.	719.	1594.
	0.162		49.4	10977.	1690.	-54.	1564.	746.	1647.
	74.	0.2265	49.4		42.35				
	671.1	0.9615	49.6	1.5138	0.2344	-0.0069	0.0413	0.0032	0.0071
38	108.3	8.0	[illegible]	11760.	1794.	-52.	271.	473.	258.
	38.4	0.0	51.0	11760.	1794.	-52.	271.	473.	246.
	0.162		49.4	12231.	1866.	-54.	282.	492.	268.
	74.	0.2265	49.7		45.07				
	671.1	0.9615	49.5	1.6942	0.2578	-0.0075	0.0074	0.0020	0.0013

TABLE V.— Continued.

RUN 9 TEST 525

PT	VKTS QPSF MTUN TEMP VSNDKTS	ALFS,U PSI RHOFIXT RHOR	JHA PED POS LONG STK LAT STK POW LEV	LIFT,U LIFT LIFT/B CL	DRAG,U DRAG DRAG/B FE CD	SIDE,U SIDE SIDE/B CY	PITCH,U PITCH PITCH/B CM	YAW,U YAW YAW/B CN	ROLL,U ROLL ROLL/B CROLL
39	108.5	[illegible]	0.0	12770.	1981.	-34.	-2232.	522.	74.
	38.1	0.0	50.4	12770.	1981.	-34.	-2232.	522.	74.
	0.162		49.5	13354.	2072.	-36.	-2124.	524.	42.
	77.	0.2273	49.6		51.48				
	673.0	0.9563	49.8	1.8511	0.2872	-0.0049	-0.0564	0.0027	0.0004
40	109.6	12.0	0.0	13855.	2233.	-52.	-2495.	574.	-858.
	38.5	0.0	50.5	13855.	2233.	-52.	-2495.	574.	-858.
	0.162		49.5	14541.	2355.	-54.	-3131.	745.	-406.
	77.	0.2273	49.6		56.35				
	673.0	0.9561	50.0	1.9915	0.3297	-0.0075	-0.0819	0.0030	-0.0034
41	108.6	13.0	0.0	14530.	2321.	-32.	-4166.	431.	-1172.
	38.2	0.0	52.3	14530.	2321.	-32.	-4166.	431.	-1172.
	0.162		49.5	14455.	2427.	-35.	-4156.	434.	-1224.
	77.	0.2273	49.6		60.75				
	673.0	0.9562	49.5	2.0679	0.3356	-0.0046	-0.1146	0.0042	-0.0053
42	108.6	0.0	0.0	2825.	338.	-20.	6749.	1649.	3083.
	38.5	0.0	50.9	2825.	338.	-20.	6749.	1649.	3083.
	0.162		49.5	2944.	475.	-21.	7644.	1721.	3148.
	76.	0.2277	49.6		41.61				
	672.2	0.9579	49.6	0.4083	0.1205	-0.0024	0.1861	0.0074	0.0137

RUN 10 TEST 525

PT	VKTS QPSF MTUN TEMP VSNDKTS	ALFS,U PSI RHOFIXT RHOR	JHA PED POS LONG STK LAT STK POW LEV	LIFT,U LIFT LIFT/B CL	DRAG,U DRAG DRAG/B FE CD	SIDE,U SIDE SIDE/B CY	PITCH,U PITCH PITCH/B CM	YAW,U YAW YAW/B CN	ROLL,U ROLL ROLL/B CROLL
4	162.6	-4.0	0.0	1500.	2125.	79.	22200.	4540.	3869.
	81.9	0.0	51.5	1500.	2125.	79.	22200.	4540.	3869.
	0.240		50.4	1451.	2265.	84.	23666.	4843.	4124.
	79.	0.2230	49.9		25.44				
	674.2	0.9321	49.9	0.0977	0.1433	0.0053	0.2865	0.0095	0.0081
5	161.0	-2.0	0.0	3910.	1989.	21.	19583.	4034.	4888.
	82.0	0.0	51.8	3910.	1989.	21.	19583.	4034.	4888.
	0.240		50.4	4184.	2125.	22.	19941.	4316.	5230.
	81.	0.2221	50.2		24.27				
	675.5	0.9336	49.9	0.2636	0.1391	0.0014	0.2346	0.0095	0.0107
6	161.2	0.0	0.0	6675.	1892.	-32.	16346.	3627.	5138.
	82.0	0.0	50.7	6675.	1892.	-32.	16346.	3627.	5138.
	0.240		50.	7156.	2028.	-34.	17204.	3888.	5504.
	82.	0.2217	49.6		23.08				
	676.1	0.9322	49.9	0.4495	0.1274	-0.0022	0.1456	0.0026	0.0108
7	161.2	2.0	0.0	9995.	1908.	-54.	10988.	3483.	7835.
	81.9	0.0	51.1	9995.	1908.	-54.	10988.	3483.	7835.
	0.240		50.3	10143.	2049.	-58.	11779.	[illegible]	8415.
	83.	0.2213	49.7		23.29				
	676.7	0.9311	49.8	0.6370	0.1287	-0.0036	0.1408	0.0073	0.0164
8	161.4	4.0	0.0	12275.	2023.	-105.	7602.	3458.	8035.
	82.0	0.0	52.8	12275.	2023.	-105.	7602.	3458.	8035.
	0.240		50.4	13104.	2173.	-113.	8165.	3714.	8630.
	83.	0.2213	49.9		24.64				
	676.7	0.9311	49.8	0.8262	0.1362	-0.0071	0.0974	0.0072	0.0168
9	161.5	6.0	0.0	15100.	2167.	-17.	2744.	3453.	10119.
	81.9	0.0	53.2	15100.	2167.	-17.	2744.	3453.	10119.
	0.240		50.5	16276.	2336.	-18.	2957.	3722.	10902.
	85.	0.2205	49.8		26.45				
	678.0	0.9277	49.9	1.0189	0.1461	-0.0011	0.0352	0.0072	0.0212
10	161.7	7.0	0.0	16390.	2338.	-38.	-40.	2482.	7598.
	81.9	0.0	50.7	16390.	2338.	-38.	-40.	2482.	7598.
	0.240		50.5	17699.	2525.	-41.	-43.	2680.	8205.
	86.	0.2201	49.8		24.54				
	678.6	0.9260	49.7	1.1056	0.1577	-0.0026	-0.0005	0.0052	0.0159
11	161.7	8.0	0.0	17565.	2462.	-20.	-491.	2220.	7304.
	82.0	0.0	52.1	17565.	2462.	-20.	-491.	2220.	7304.
	0.240		50.4	18966.	2659.	-22.	-1110.	2318.	7880.
	86.	0.2201	49.8		30.04				
	678.6	0.9260	49.8	1.1890	0.1662	-0.0013	-0.0127	0.0046	0.0155
12	141.7	-4.0	50.0	6765.	2924.	-112.	22654.	1525.	3606.
	63.3	0.0	49.8	6765.	2924.	-112.	22654.	1525.	3606.
	0.210		50.3	7252.	3137.	-120.	24302.	1636.	4182.
	86.	0.2216	49.8		48.10				
	678.6	0.9322	49.8	0.5898	0.2551	-0.0098	0.3761	0.0041	0.0103
13	141.5	-2.0	50.0	8805.	3069.	-113.	20341.	1793.	2342.
	63.1	0.0	51.2	8805.	3069.	-113.	20341.	1793.	2342.
	0.209		50.3	9512.	3296.	-121.	21954.	1927.	2517.
	87.	0.2212	49.7		49.64				
	679.2	0.9306	49.7	0.7587	0.2667	-0.0099	0.3384	0.0049	0.0054
14	141.7	0.0	50.0	10740.	3227.	-136.	16777.	2048.	[illegible]
	63.1	0.0	53.3	10740.	3227.	-136.	16777.	2048.	[illegible]
	0.209		50.5	11502.	3474.	-146.	18062.	2205.	3240.
	88.	0.2208	49.8		51.14				
	679.9	0.9289	49.8	0.9251	0.2825	-0.0119	0.2795	0.0057	0.0082
15	141.0	2.0	50.0	12590.	3487.	-152.	13495.	2071.	3647.
	63.1	0.0	51.6	12590.	3487.	-152.	13495.	2071.	3647.
	0.209		50.4	13554.	3754.	-164.	14741.	2223.	3950.
	88.	0.2208	49.8		55.30				
	679.9	0.9289	49.0	1.0351	0.3055	-0.0133	0.2263	0.0056	0.0101

TABLE V.— Continued.

RUN 10 TEST 525

PT	VKTS UPSF MACH TEMP VSRORTS	ALFS,C PSI RHO*10 RHOR	INA PED POS LONG STK LAT STK RUD DEV	LIFT,C LIFT LIFT/6 CL	DRAG,C DRAG DRAG/6 FE CD	SIDE,C SIDE SIDE/6 CY	PITCH,C PITCH PITCH/6 CM	YAW,C YAW YAW/6 CN	ROLL,C ROLL ROLL/6 CROLL
16	101.8	44.0	75.0	2565.	2187.	+41.	11819.	332.	415.
	32.8	0.0	52.0	2585.	2187.	+41.	11819.	332.	415.
	0.150		50.5	2754.	2330.	+42.	12591.	354.	442.
	86.	0.2231	49.9		66.72				
	679.9	0.9358	49.8	0.4337	0.3686	+0.0014	0.3791	0.0017	0.0022
17	101.5	+2.0	75.0	3565.	2184.	+61.	10946.	474.	641.
	32.9	0.0	51.1	3565.	2184.	+60.	10944.	474.	640.
	0.150		50.6	3720.	2314.	+64.	[illegible]	503.	680.
	86.	0.2234	49.5		66.44				
	679.6	0.9422	49.8	0.5097	0.3671	+0.0091	0.3501	0.0025	0.0033
18	101.3	0.0	75.0	4480.	2224.	-44.	10714.	224.	207.
	32.8	0.0	52.0	4480.	2224.	-44.	10714.	224.	207.
	0.150		51.7	4733.	2360.	+45.	11373.	237.	220.
	86.	0.2240	49.8		67.40				
	678.6	0.9422	49.8	0.7523	0.3751	-0.0067	0.3640	0.0012	0.0011
19	101.6	1.0	75.0	4950.	2275.	-72.	9673.	137.	737.
	32.9	0.0	52.0	4950.	2275.	-72.	9673.	137.	737.
	0.150		50.5	5201.	2415.	+76.	10267.	144.	804.
	86.	0.2234	49.6		65.10				
	678.6	0.9427	50.1	0.8227	0.3821	-0.0121	0.3092	0.0019	0.0040
20	119.1	44.0	60.0	4200.	2854.	-39.	17567.	285.	244.
	45.0	0.0	49.8	4200.	2854.	-39.	17567.	285.	244.
	0.176		50.5	4477.	3042.	+42.	18703.	304.	400.
	86.	0.2235	49.5		65.40				
	679.6	0.9382	49.9	0.5155	0.3505	-0.0048	0.4048	0.0031	0.0032
21	119.0	+2.0	60.0	5420.	2914.	+46.	16962.	572.	495.
	45.0	0.0	50.0	5420.	2914.	+46.	16962.	572.	495.
	0.176		50.6	5766.	3100.	+49.	18046.	608.	462.
	85.	0.2234	49.6		66.70				
	678.7	0.9399	49.9	0.6649	0.3578	-0.0056	0.3959	0.0022	0.0036
22	118.8	+1.0	60.0	6105.	2937.	+66.	16474.	995.	740.
	45.0	0.0	50.0	6105.	2937.	+66.	16474.	995.	740.
	0.176		50.7	6495.	3114.	+70.	17526.	1058.	788.
	85.	0.2234	49.9		65.21				
	679.0	0.9400	49.9	0.7500	0.3623	-0.0081	0.3851	0.0039	0.0028

RUN 11 TEST 525

PT	VKTS UPSF MACH TEMP VSRORTS	ALFS,C PSI RHO*10 RHOR	INA PED POS LONG STK LAT STK RUD DEV	LIFT,C LIFT LIFT/6 CL	DRAG,C DRAG DRAG/6 FE CD	SIDE,C SIDE SIDE/6 CY	PITCH,C PITCH PITCH/6 CM	YAW,C YAW YAW/6 CN	ROLL,C ROLL ROLL/6 CROLL
3	100.3	+4.0	75.0	2525.	2108.	-27.	11276.	595.	568.
	31.5	0.0	52.1	2525.	2108.	-27.	11276.	595.	568.
	0.149		49.8	2727.	2273.	-29.	12190.	647.	613.
	85.	0.2231	49.4		66.74				
	676.7	0.9258	50.1	0.4925	0.3697	-0.0047	0.3766	0.0033	0.0031
4	100.4	+2.0	75.0	3395.	2108.	-65.	10573.	741.	645.
	31.6	0.0	51.1	3395.	2108.	-65.	10573.	741.	645.
	0.149		49.8	3667.	2275.	-65.	11421.	801.	697.
	85.	0.2266	50.1		66.64				
	676.7	0.9258	50.1	0.5941	0.3692	-0.0105	0.3521	0.0044	0.0044
5	100.3	0.0	75.0	4255.	2148.	-60.	9927.	240.	270.
	31.5	0.0	50.0	4255.	2148.	-60.	9927.	240.	270.
	0.149		49.9	4595.	2320.	-67.	[illegible]	263.	291.
	85.	0.2261	50.0		67.13				
	676.7	0.9258	50.0	0.7457	0.3760	-0.0109	0.3327	0.0013	0.0015
6	163.4	2.0	0.0	8555.	2111.	-1374.	10972.	17751.	4624.
	82.0	-2.0	50.2	8555.	2111.	-1374.	10972.	17751.	4624.
	0.243		49.8	9425.	2320.	-1514.	12092.	19562.	5096.
	84.	0.2157	50.0		25.14				
	677.4	0.9074	50.0	0.5750	0.1422	-0.0926	0.1467	0.2359	[illegible]
7	163.5	2.0	0.0	8665.	2046.	-672.	11708.	11384.	8172.
	81.7	+2.0	50.0	8665.	2046.	-672.	11708.	11384.	8172.
	0.242		49.5	9545.	2207.	-745.	12473.	12613.	8939.
	87.	0.2145	50.2		25.05				
	679.2	0.9025	50.0	0.5858	0.1380	-0.0455	0.1507	0.0234	0.0110
8	163.2	2.0	0.0	8225.	2027.	610.	13230.	-1121.	7232.
	81.3	2.0	50.7	8225.	2027.	610.	13230.	-1121.	7232.
	0.242		49.8	9124.	2254.	677.	14684.	-2021.	8027.
	85.	0.2142	50.0		24.94				
	679.9	0.9015	50.1	0.5591	0.1378	0.0415	0.0711	[illegible]	[illegible]
9	163.4	2.0	0.0	7950.	2035.	1294.	14032.	-7528.	8379.
	81.5	0.0	50.9	7950.	2035.	1294.	14032.	-7528.	8379.
	0.242		49.8	8817.	2263.	1440.	15647.	-8371.	9317.
	84.	0.2138	50.0		25.23				
	680.5	0.8993	50.0	0.5388	0.1379	0.0877	[illegible]	-0.0159	0.0177
10	163.3	2.0	0.0	7700.	2087.	-18653.	16113.	-12651.	54633.
	81.1	9.5	51.0	7700.	2087.	-18653.	16113.	-12651.	54633.
	0.241		49.8	8577.	2325.	-20755.	17947.	-14083.	60853.
	90.	0.2150	49.4		25.78				
	681.1	0.8978	49.6	0.5209	0.1422	-1.2649	0.2096	[illegible]	1.2537
11	163.6	2.0	0.0	7370.	2083.	30.	33649.	3305.	7521.
	81.2	0.0	49.9	7370.	2083.	30.	33649.	3305.	7521.
	0.241		49.3	8224.	2302.	33.	37374.	3755.	8444.
	91.	0.2136	49.7		25.42				
	681.7	0.8961	50.6	0.5016	0.1414	0.0020	0.2359	[illegible]	[illegible]

TABLE V.— Continued.

RUN 11 TEST 525

PT	KTS QPSF MION TEMP V3NDKT3	ALPHA,C PSI RHO/10 RHOW	INA REC RMS LONG S P LAT S P PUM LEV	LIFT,C LIFT LIFT/Q CL	DRAG,C DRAG DRAG/Q FE CD	SIDE,C SIDE SIDE/Q CY	PITCH,C PITCH PITCH/Q CM	YAW,C YAW YAW/Q CN	ROLL,C ROLL ROLL/Q CROLL
12	163.5	2.0	0.0	9765.	2068.	-37.	-12279.	4509.	6436.
	81.1	0.0	49.6	9765.	2068.	-37.	-12279.	4509.	6436.
	0.241		49.0	10897.	2306.	-41.	-13792.	5052.	7182.
	91.	0.2130	50.9		25.44				
	691.7	0.8951	50.0	0.6650	0.1408	-0.0025	-0.0591	0.0095	0.0136
13	153.6	2.0	0.0	8346.	2083.	-69.	11358.	6860.	18739.
	91.0	0.0	50.2	8387.	2083.	-69.	11358.	6862.	18739.
	0.241		49.6	9306.	2324.	-77.	12674.	7432.	20909.
	91.	0.2130	54.6		25.71				
	681.7	0.8962	50.0	0.9888	0.1421	-0.0047	0.0774	0.0141	0.0397
14	163.3	2.0	0.0	8135.	2195.	-7.	11361.	6371.	26625.
	90.8	0.0	51.0	8135.	2195.	-7.	11361.	6371.	26625.
	0.241		49.6	9093.	2454.	-8.	12700.	4357.	31326.
	92.	0.2126	50.7		27.17				
	682.3	0.8946	49.9	0.5565	0.1501	-0.0005	0.1474	0.0176	0.0590
15	163.5	2.0	0.0	8445.	2055.	-420.	11960.	14556.	2925.
	81.0	0.0	-1709.0	8445.	2055.	-420.	11966.	14556.	2925.
	0.241		49.8	9440.	2297.	-470.	13376.	16274.	3270.
	92.	0.2126	49.7		25.38				
	682.3	0.8906	49.9	0.5755	0.1402	-0.0289	0.1554	0.0309	0.0062
16	163.5	2.0	0.0	8380.	2092.	-754.	12535.	23590.	2615.
	81.0	0.0	-1709.0	8380.	2092.	-754.	12535.	23590.	2615.
	0.241		49.9	9366.	2339.	-843.	14012.	26370.	2923.
	92.	0.2126	49.6		25.85				
	682.3	0.8946	50.0	0.5717	0.1427	-0.0514	0.1627	0.0500	0.0055
17	122.3	8.0	0.0	8790.	1327.	-7.	-8615.	1321.	3108.
	96.0	0.0	48.2	8790.	1327.	-7.	-8615.	1321.	3108.
	0.180		44.8	9667.	1459.	-8.	-8946.	1453.	3404.
	90.	0.2161	49.5		28.84				
	681.7	0.9093	50.0	1.0555	0.1592	-0.0008	-0.0165	0.0049	0.0116
18	122.2	8.0	0.0	8710.	1352.	21.	-9985.	1206.	10025.
	96.0	0.0	48.7	8710.	1352.	21.	-9985.	1206.	10025.
	0.180		44.8	9579.	1487.	25.	-10983.	1327.	11085.
	90.	0.2161	54.5		29.42				
	681.1	0.9093	50.0	1.0470	0.1625	0.0026	-0.0225	0.0045	0.0397
19	122.1	0.0	0.0	8605.	1409.	44.	-14260.	1465.	16545.
	45.9	0.0	49.8	8605.	1409.	49.	-14260.	1465.	16545.
	0.180		49.8	9461.	1549.	54.	-15684.	1611.	18194.
	90.	0.2161	58.5		30.69				
	681.1	0.9093	50.0	1.0359	0.1695	0.0059	-0.0326	0.0055	0.0619
20	121.7	12.0	0.0	11210.	1673.	-13.	-4079.	1104.	3100.
	45.6	0.0	49.0	11210.	1673.	-13.	-4079.	1104.	3100.
	0.179		49.8	12326.	1840.	-14.	-4485.	1213.	3415.
	90.	0.2162	49.4		36.66				
	681.1	0.9094	49.9	1.3577	0.2026	-0.0016	-0.0960	0.0042	0.0117
21	121.7	12.0	0.0	11120.	1695.	66.	-3758.	-607.	4838.
	45.6	0.0	49.5	11120.	1695.	66.	-3758.	-607.	4838.
	0.179		50.0	12227.	1864.	73.	-4132.	-555.	[illegible]
	90.	0.2162	64.3		37.16				
	681.1	0.9094	50.0	1.3466	0.2053	0.0080	-0.1866	-0.0018	0.0570
22	121.9	12.0	0.0	11035.	1744.	135.	-6507.	-1583.	15558.
	95.7	0.0	49.0	11035.	1744.	135.	-6507.	-1583.	15558.
	0.180		49.5	12156.	1921.	149.	-7244.	-1744.	17137.
	91.	0.2158	80.6		36.17				
	681.7	0.9078	50.0	1.3385	0.2109	0.0163	-0.1991	-0.0061	0.0584
23	161.5	0.0	0.0	13140.	2906.	-1455.	20121.	15479.	900.
	79.2	-4.0	49.5	13140.	2906.	-1455.	20121.	15479.	900.
	0.238		49.9	14652.	3240.	-1622.	22437.	17261.	1070.
	91.	0.2132	49.4		36.70				
	681.7	0.8968	50.0	0.9169	0.2028	-0.1015	0.2672	0.0356	0.0021
24	161.5	0.0	0.0	13180.	2847.	-834.	19766.	8372.	-80.
	79.2	-2.0	49.2	13180.	2847.	-834.	19768.	8372.	-80.
	0.238		49.5	14697.	3175.	-930.	22045.	9336.	-89.
	91.	0.2132	49.4		35.47				
	681.7	0.8968	49.9	0.9199	0.1987	-0.0585	0.2626	0.0182	-0.0002
25	161.3	0.0	0.0	13045.	2795.	377.	19328.	-3193.	2595.
	78.9	2.0	49.8	13099.	2795.	377.	19328.	-3193.	2595.
	0.235		49.4	14571.	3116.	421.	21588.	-3566.	2898.
	92.	0.2128	49.3		35.40				
	682.3	0.8953	50.0	0.9144	0.1958	0.0264	0.2578	-0.0076	0.0357
26	161.2	0.0	0.0	13065.	2813.	1077.	19632.	-4456.	4331.
	78.7	4.0	49.0	13065.	2813.	1077.	19632.	-4456.	4331.
	0.235		50.1	14593.	3142.	1203.	22151.	[illegible]	4838.
	92.	0.2128	49.5		[illegible]				
	682.3	0.8953	49.9	0.9170	0.1974	0.0756	0.2644	-0.0206	0.0096
27	161.1	0.0	0.0	12790.	2905.	2333.	19845.	-21605.	7985.
	78.5	6.0	48.9	12790.	2905.	2333.	19845.	-21605.	7985.
	0.237		49.2	14254.	3245.	2610.	22262.	-24113.	8932.
	93.	0.2124	49.6		36.43				
	683.0	0.8938	50.0	0.8963	0.2009	0.1645	0.2657	-0.0473	0.0175
28	161.4	0.0	0.0	12595.	2815.	-164.	35507.	2495.	2547.
	78.8	0.0	49.5	12595.	2815.	-164.	35507.	2495.	2547.
	0.235		32.0	13972.	3153.	-184.	39744.	2754.	2852.
	93.	0.2124	49.8		35.75				
	683.0	0.8937	50.1	0.8868	0.1975	-0.0115	0.4744	0.0054	0.0055
29	161.4	0.0	0.0	14215.	2833.	-191.	[illegible]	[illegible]	2436.
	79.8	0.0	44.7	14215.	2833.	-191.	[illegible]	[illegible]	2436.
	0.236		69.6	15906.	3170.	-214.	[illegible]	[illegible]	3283.
	93.	0.2124	49.1		35.45				
	683.0	0.8937	49.9	0.9969	0.1987	-0.0136	-0.0522	[illegible]	0.0054

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TABLE V.— Continued.

RUN 11 TEST 525

PT	VKTS QPSF RTUN TEMP VSNDKTS	ALFS,C PSI MAJOR LOC MACH	INA FED POS LONG STN LAT STN POW LEV	LIFT,C LIFT LIFT/B CL	DRAG,C DRAG DRAG/B FE CD	SIDE,C SIDE SIDE/B CY	PITCH,C PITCH PITCH/C CM	YAW,C YAW YAW/B CN	ROLL,C ROLL ROLL/B CROLL
30	161.3	0.0	[illegible]	12765.	2776.	-346.	20018.	1699.	12285.
	78.7	0.0	49.7	12765.	2776.	-348.	20018.	1699.	12286.
	0.235		49.[illegible]	14362.	3106.	-399.	22399.	1901.	13750.
	93.	0.2124	49.2		35.27				
	683.0	0.8937	49.8	0.8977	0.1928	-0.0244	[illegible]	0.0037	[illegible]
31	161.4	0.0	[illegible]	12500.	2776.	-442.	21422.	2594.	21594.
	78.7	0.0	49.0	12500.	2777.	-442.	21420.	2594.	21594.
	0.236		49.[illegible]	13700.	3105.	-495.	24077.	2913.	24276.
	94.	0.2121	48.5		35.26				
	683.6	0.8921	49.0	0.8635	0.1925	-0.0310	0.2922	[illegible]	[illegible]
32	161.6	[illegible]	[illegible]	13215.	2823.	-653.	19773.	13975.	-812.
	78.8	0.0	-179.0	13215.	2823.	-653.	19773.	13975.	-812.
	0.238		49.8	14814.	3165.	-732.	22162.	15665.	-910.
	94.	0.2124	49.8		35.82				
	683.6	0.8920	50.0	0.9263	0.1979	-0.0458	0.2057	[illegible]	[illegible]
33	161.5	0.0	[illegible]	13210.	2844.	-997.	19174.	22585.	-2353.
	78.7	0.0	+179.0	13210.	2844.	-997.	19178.	22586.	-2353.
	0.238		49.8	14808.	3188.	-1118.	21498.	25249.	-2638.
	94.	0.2120	49.6		36.18				
	683.6	0.8921	49.9	0.9269	0.1999	-0.0700	0.2561	0.0092	-0.0051
34	112.7	0.0	[illegible]	11105.	1616.	-57.	455.	814.	840.
	38.9	0.0	50.5	11149.	1818.	-52.	469.	814.	840.
	0.165		49.7	12355.	2027.	-57.	514.	900.	920.
	94.	0.2151	49.7		46.70				
	683.6	0.9050	49.9	1.5572	0.2550	-0.0074	0.0127	0.0056	0.0037
35	112.5	0.0	0.0	10975.	1836.	-169.	901.	-362.	4577.
	38.9	0.0	51.0	10975.	1835.	-169.	901.	-362.	4577.
	0.165		49.6	12083.	2021.	-186.	992.	-400.	5039.
	92.	0.2159	49.2		47.19				
	682.3	0.9023	50.0	1.5564	0.2602	-0.0240	0.0243	-0.0003	0.0292
36	112.7	0.0	[illegible]	10065.	1771.	-216.	2592.	-744.	10565.
	39.0	0.0	50.5	10065.	1771.	-216.	2594.	-744.	10566.
	0.166		49.5	11764.	1952.	-238.	2860.	-819.	11635.
	92.	0.2154	50.5		45.35				
	682.3	0.9083	50.0	1.5178	0.2505	-0.0306	0.0699	-0.0105	0.0465

RUN 12 TEST 525

PT	VKTS QPSF RTUN TEMP VSNDKTS	ALFS,C PSI MAJOR LOC MACH	INA FED POS LONG STN LAT STN POW LEV	LIFT,C LIFT LIFT/B CL	DRAG,C DRAG DRAG/B FE CD	SIDE,C SIDE SIDE/B CY	PITCH,C PITCH PITCH/C CM	YAW,C YAW YAW/B CN	ROLL,C ROLL ROLL/B CROLL
3	80.1	-12.0	90.0	-825.	1522.	-14.	7005.	-433.	-151.
	21.4	0.0	49.0	-825.	1522.	-14.	7005.	-433.	-151.
	0.121		50.0	-837.	1545.	-14.	7002.	-434.	-155.
	65.	0.2341	49.9		71.13				
	685.9	0.9851	50.0	-0.2150	0.1930	-0.0036	0.3776	0.0035	-0.0012
4	80.3	-8.0	90.0	265.	1427.	-74.	6844.	-466.	-243.
	21.5	0.0	49.3	265.	1427.	-74.	6849.	-466.	-243.
	0.121		50.[illegible]	269.	1449.	-75.	6745.	-473.	-246.
	65.	0.2341	50.2		86.32				
	685.4	0.9851	50.1	0.0690	0.3669	-0.0190	0.3240	0.0037	-0.0023
5	80.4	-4.0	90.0	1445.	1395.	-45.	5956.	-425.	-175.
	21.5	0.0	47.8	1445.	1395.	-45.	5956.	-426.	-175.
	0.121		50.1	1467.	1416.	-49.	6040.	-435.	-178.
	65.	0.2341	49.9		96.70				
	685.4	0.9857	50.[illegible]	0.3700	0.3570	-0.0251	0.2907	0.0030	-0.0014
6	80.5	-2.0	90.0	2045.	1420.	-74.	6301.	176.	204.
	21.5	0.0	49.0	2045.	1420.	-74.	6301.	176.	204.
	0.121		50.0	2076.	1442.	-75.	6450.	179.	206.
	65.	0.2341	49.9		65.77				
	685.0	0.9850	50.0	0.5233	0.3659	-0.0189	0.3098	0.0014	0.0016
7	80.4	0.0	90.0	2605.	1441.	-50.	5587.	216.	126.
	21.5	0.0	49.7	2605.	1441.	-50.	5587.	216.	126.
	0.121		50.1	2705.	1463.	-51.	5671.	220.	128.
	65.	0.2341	49.7		[illegible]				
	685.9	0.9851	50.0	0.6655	0.3590	-0.0126	0.2720	0.0017	0.0010
8	80.5	2.0	90.0	3265.	1492.	-40.	4916.	-45.	-125.
	21.6	0.0	48.0	3265.	1492.	-40.	4916.	-45.	-125.
	0.121		50.1	3341.	1514.	-41.	5002.	-45.	-127.
	65.	0.2337	50.0		89.15				
	686.1	0.9832	50.1	0.8278	0.3622	-0.0102	0.2344	0.0007	-0.0001
9	80.8	4.0	90.0	3530.	1544.	-13.	4150.	[illegible]	-171.
	21.5	0.0	48.0	3530.	1545.	-13.	4150.	[illegible]	-171.
	0.121		50.0	3595.	1574.	-13.	4221.	32.	-174.
	65.	0.2337	49.9		71.96				
	686.1	0.9832	50.1	0.9058	0.3975	-0.0033	0.2028	0.0025	-0.0014
10	80.9	8.0	90.	5005.	1735.	-13.	2189.	294.	201.
	21.5	0.0	48.6	5005.	1735.	-13.	2189.	[illegible]	201.
	0.121		50.0	5091.	1764.	-13.	2227.	299.	[illegible]
	66.	0.2337	49.8		[illegible]				
	686.1	0.9832	50.0	1.2850	0.4473	-0.0033	0.1077	0.0025	0.0016
11	80.8	12.0	90.0	6155.	1925.	-5.	73.	474.	-76.
	21.5	0.0	48.8	6155.	1925.	-5.	73.	474.	-76.
	0.121		50.5	6250.	1954.	-5.	74.	488.	-77.
	66.	0.2337	49.9		89.30				
	686.1	0.9832	50.0	1.5827	0.4950	-0.0013	0.0056	0.0058	-0.0006

TABLE V.— Continued.

RUN 12 TEST 525

PT	VKTS QPSF MACH TEMP VSQRTS	ALFS,L PSI RNO*100 RNOW	INA PED POS LONG STK LAT STK POW CEY	LIFT,U LIFT LIFT/6 CL	DRAG,U DRAG DRAG/6 FE CD	SIDE,U SIDE SIDE/6 CY	PITCH,U PITCH PITCH/6 CM	YAW,U YAW YAW/6 CN	ROLL,U ROLL ROLL/6 CROLL
12	60.4	13.0	40.0	6410.	1973.	-4.	-667.	532.	234.
	21.5	0.0	47.6	6010.	1973.	-4.	-667.	532.	234.
	0.12		50.0	6532.	2011.	-4.	-563	544.	236.
	61.	0.2333	49.9		91.99				
	666.7	0.9813	50.0	1.6995	0.5077	+0.0023	-0.0424	0.0051	0.0016
13	60.2	14.0	40.0	6045.	2250.	-109.	-5866.	-404.	-2712.
	21.6	0.0	47.2	6045.	2256.	-109.	-5866.	-403.	-2712.
	0.120		50.0	6160.	2299.	-111.	-5944.	-417.	-2763.
	61.	0.2333	49.9		105.61				
	666.7	0.9814	50.0	1.5643	0.5839	+0.0266	-1.1494	-0.0072	-0.0418
14	59.6	-4.0	75.0	2645.	2215.	-53.	10733.	5.	69.
	32.8	0.0	48.5	2645.	2215.	-53.	10733.	4.0	69.
	0.150		50.1	2711.	2270.	-54.	10999.	5.	71.
	64.	0.2319	49.8		67.57				
	667.3	0.9757	50.0	0.3058	0.3733	-0.0069	0.3453	0.0002	0.0004
15	99.6	-2.0	75.0	3590.	2183.	-97.	9675.	475.	1043.
	32.7	0.0	47.8	3590.	2183.	-97.	9675.	475.	1043.
	0.150		50.0	3666.	2242.	-100.	10141.	488.	1071.
	69.	0.2315	50.2		66.74				
	668.0	0.9739	50.7	0.5088	0.3589	+0.0168	0.3175	0.0025	0.0055
16	99.9	0.0	75.0	4500.	2257.	-99.	9142.	465.	376.
	32.9	0.0	47.0	4500.	2257.	-99.	9142.	465.	376.
	0.150		50.1	4621.	2310.	-102.	9350.	480.	386.
	69.	0.2315	49.9		86.01				
	668.0	0.9738	49.9	0.7998	0.3791	-0.0166	0.2922	0.0024	0.0020
17	99.9	2.0	75.0	5590.	2337.	-86.	8063.	349.	-360.
	32.8	0.0	47.0	5590.	2337.	-84.	8063.	349.	-360.
	0.150		50.1	5545.	2404.	-91.	8295.	359.	-370.
	73.	0.2314	49.9		71.19				
	666.6	0.9727	50.0	0.9072	0.3953	-0.0146	0.2582	0.0016	-0.0019
18	99.8	4.0	75.0	6245.	2441.	-74.	6497.	195.	-124.
	32.8	0.0	47.2	6245.	2441.	-74.	6497.	195.	-124.
	0.150		49.9	6425.	2511.	-76.	6684.	201.	-128.
	73.	0.2313	50.0		74.42				
	665.8	0.9720	50.0	1.0578	0.4111	-0.0125	0.2052	0.0061	-0.0046

RUN 13 TEST 525

PT	VKTS QPSF MACH TEMP VSQRTS	ALFS,L PSI RNO*100 RNOW	INA PED POS LONG STK LAT STK POW CEY	LIFT,U LIFT LIFT/6 CL	DRAG,U DRAG DRAG/6 FE CD	SIDE,U SIDE SIDE/6 CY	PITCH,U PITCH PITCH/6 CM	YAW,U YAW YAW/6 CN	ROLL,U ROLL ROLL/6 CROLL
3	100.3	-4.0	75.0	1715.	1992.	-20.	11296.	851.	1054.
	32.9	0.0	49.3	1715.	1992.	-20.	11296.	851.	1054.
	0.150		50.0	1776.	2062.	-21.	11696.	880.	1091.
	73.	0.2296	49.8		60.60				
	670.9	0.9659	50.3	0.2892	0.3360	-0.0034	0.3613	0.0042	0.0055
4	100.3	-2.0	75.0	2470.	1978.	-4.	10265.	771.	554.
	32.9	0.0	49.7	2470.	1978.	-4.	10265.	771.	554.
	0.150		49.9	2557.	2048.	-4.	10626.	805.	584.
	73.	0.2296	49.6		60.17				
	670.9	0.9658	50.0	0.4157	0.3329	-0.0007	0.3281	0.0041	0.0045
5	100.3	0.0	75.0	3385.	1984.	-91.	9863.	1019.	541.
	32.9	0.0	49.3	3385.	1984.	-91.	9863.	1019.	541.
	0.150		50.0	3505.	2056.	-94.	10192.	1050.	560.
	73.	0.2296	49.7		60.37				
	670.9	0.9658	50.0	0.5685	0.3335	-0.0153	0.3146	0.0053	0.0028
6	100.3	2.0	75.0	4215.	2054.	-69.	7697.	555.	637.
	32.8	0.0	51.1	4215.	2054.	-69.	7697.	555.	637.
	0.150		49.9	4372.	2130.	-72.	7964.	576.	661.
	74.	0.2291	49.6		62.53				
	671.1	0.9640	50.0	0.7089	0.3455	-0.0116	0.2463	0.0029	0.0033
7	100.4	4.0	75.0	4235.	2005.	-15.	5147.	143.	438.
	32.9	0.0	49.8	4235.	2005.	-15.	5147.	143.	438.
	0.150		50.0	4393.	2080.	-16.	5462.	148.	475.
	74.	0.2291	49.8		60.95				
	671.1	0.9640	50.0	0.7113	0.3367	-0.0025	0.1857	0.0007	0.0044
8	100.5	6.0	75.0	3305.	1925.	-58.	4633.	446.	759.
	33.0	0.0	65.9	3305.	1925.	-57.	4633.	446.	759.
	0.150		50.1	3425.	1997.	-60.	4806.	462.	788.
	74.	0.2291	50.4		56.55				
	671.1	0.9640	50.0	0.5535	0.3229	-0.0097	0.1477	0.0049	0.0080
9	100.5	8.0	75.0	4165.	1955.	-22.	3457.	696.	1178.
	33.0	0.0	50.0	4165.	1955.	-22.	3457.	696.	1178.
	0.150		50.0	4321.	2026.	-23.	3586.	722.	1222.
	74.	0.2291	49.3		59.30				
	671.1	0.9640	49.9	0.6990	0.3276	-0.0037	0.1103	0.0036	0.0061
10	100.4	8.0	75.0	4865.	2054.	12.	2127.	606.	477.
	32.9	0.0	50.7	4865.	2054.	12.	2127.	606.	477.
	0.150		51.0	5047.	2131.	12.	[illegible]	629.	[illegible]
	74.	0.2291	49.8		62.48				
	671.1	0.9640	50.0	0.8175	0.3452	0.0020	0.0684	0.0032	0.0051
11	99.4	0.0	45.0	5710.	1940.	23.	1665.	139.	1541.
	32.5	0.0	49.5	5710.	1940.	23.	1665.	139.	1541.
	0.149		50.3	5933.	2016.	24.	1730.	143.	1593.
	75.	0.2287	50.0		54.61				
	671.7	0.9623	50.0	0.9632	0.3294	0.0039	0.0544	0.0060	0.0071
12	100.4	6.0	30.0	6045.	1669.	23.	1256.	545.	1440.
	32.7	0.0	55.5	5575.	1669.	23.	1256.	545.	1444.
	0.149		50.0	6202.	1729.	24.	1301.	573.	1491.
	75.	0.2287	49.5		54.45				
	671.7	0.9621	50.0	1.0226	0.2815	0.0039	0.0405	0.0094	0.0097
13	100.5	0.0	[illegible]	6700.	945.	-4.	-1487.	622.	2045.
	33.0	0.0	49.8	6700.	905.	-4.	-1487.	622.	2045.
	0.150		49.9	6963.	982.	-4.	-1540.	654.	2104.
	75.	0.2287	49.4		30.44				
	671.7	0.9622	49.5	1.1225	0.1583	-0.0015	-0.0472	0.0043	0.0136

TABLE V.— Continued.

RUN 15 TEST 525

PT											
7	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
8	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
9	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
10	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
11	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
12	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
13	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
14	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
15	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
16	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
17	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
18	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
19	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
20	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
21	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
22	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
23	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
24	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]

TABLE V.— Continued.

RUN [illegible] TEST 525

P	VKTS GPSF MACH TEMP VSNDKTS	ALFS,C PSI [illegible] ARDF	INA REQ POS LONG STA LAT STA POR LEV	LIFT,L LIFT LIFT/Q CL	DRAG,L DRAG DRAG/Q FE CD	SIDE,L SIDE SIDE/Q CY	PITCH,L PITCH PITCH/Q CM	YAW,L YAW YAW/Q CN	ROLL,L ROLL ROLL/Q CROLL	RPM OMEGAR R/DW	HHAST,L CHAST,L HP HP/Q CP/SIGMA
[illegible]	154.6 65.1 0.209 71. 659.2	0.0 0.0 0.2277 0.9569	0.0 50.3 44.8 50.0 10.6	4435. 4555. 4646. 0.3899	654. 654. 686. 10.58 0.0572	-7. -7. -7. -0.0050	9694. 9694. 9453. 0.1515	5532. 5532. 5754. 0.0150	5910. 5910. 5970. 0.0153	450.2 547. 0.344	25720. 36830. 454.2 475.7 0.005284
[illegible]	154.6 79.7 0.235 93. 679.9	0.0 0.0 0.2196 0.9520	0.0 50.4 50.0 44.6 14.5	5695. 5695. 6163. 0.3950	934. 934. 1011. 11.72 0.0678	-29. -29. -31. -0.0026	12472. 12472. 13497. 0.1546	7313. 7313. 7915. 0.0154	3542. 3542 3833. 0.0076	455.7 606. 0.434	55237. 49166. 544.7 649.1 0.007929
5	159.5 79.5 0.216 89. 680.5	0.0 0.0 0.2192 0.9724	0.0 50.0 50.0 44.6 14.8	5810. 5810. 6294. 0.4039	791. 791. 856. 9.95 0.0550	-20. -20. -22. -0.0014	12156. 12156. 13179. 0.1606	6741. 6741. 7303. 0.0147	2455. 2455. 3600. 0.0064	455.2 532. 0.444	37471. 52172. 655.3 710.5 0.004780
6	177.4 95.6 0.260 101. 687.9	0.0 0.0 0.2132 0.4971	0.0 50.7 50.0 45.2 19.7	7530. 7530. 8394. 0.4352	1010. 1010. 1126. 10.57 0.0580	8. 8. 9. 0.0005	14313. 14313. 15955. 0.1574	2944. 2944. 3282. 0.0053	5093. 5093. 5700. [illegible]	458.1 600. 0.444	56524. 57715. 645.3 442.3 0.011586
7	178.5 95.6 0.266 106. 692.2	0.0 0.0 0.2106 0.8860	0.0 50.8 50.0 45.4 26.0	7600. 7600. 8577. 0.4392	504. 504. 569. 5.27 0.0291	-55. -55. -60. -0.0031	14163. 14163. 16007. 0.1560	4571. 4571. 5159. 0.0082	4347. 4347. 4903. 0.0079	459.4 601. 0.511	70261. 70892. 1131.1 1276.6 0.015522
8	179.3 96.0 0.266 110. 693.6	0.0 0.0 0.2096 0.8828	0.0 50.7 50.0 45.5 39.4	7655. 7655. 8671. 0.4403	-202. -202. -229. -2.10 -0.0116	-16. -16. -18. -0.0009	12710. 12710. 14397. 0.1391	4241. 4241. 4800. 0.0077	5071. 5071. 5745. 0.0093	459.9 601. 0.509	102141. 100939. 1522.3 1724.5 0.021082

RUN 1[illegible] TEST 525

P	VKTS GPSF MACH TEMP VSNDKTS	ALFS,C PSI [illegible] ARDF	INA REQ POS LONG STA LAT STA POR LEV	LIFT,L LIFT LIFT/Q CL	DRAG,L DRAG DRAG/Q FE CD	SIDE,L SIDE SIDE/Q CY	PITCH,L PITCH PITCH/Q CM	YAW,L YAW YAW/Q CN	ROLL,L ROLL ROLL/Q CROLL	RPM OMEGAR R/DW	HHAST,L CHAST,L HP HP/Q CP/SIGMA
4	169.5 38.3 0.102 84. 677.4	0.0 0.0 0.2244 0.9442	0.0 50.8 50.2 44.0 13.0	6635. 6635. 7077. 0.9559	395. 395. 418. 10.30 0.0569	-81. -81. -86. -0.0117	5969. 5969. 6322. 0.1036	1663. 1663. [illegible] 0.0083	1156. 1156. 1224. 0.0052	564.4 739. 0.250	24200. 27540. 404.0 491.4 0.003222
5	150.3 54.0 0.193 84. 677.4	0.0 0.0 0.2232 0.9390	0.0 50.1 50.1 49.0 16.2	9480. 9480. 10096. 0.9698	375. 375. 399. 6.94 0.0384	-163. -163. -170. -0.0164	7597. 7597. 8091. 0.1479	1665. 1665. 1923. 0.0057	890. 890. 954. 0.0028	564.2 734. 0.296	43360. 42374. 767.0 810.8 0.005561
6	149.0 69.2 0.220 92. 682.3	0.0 0.0 0.2167 0.9243	0.0 50.2 50.0 44.2 22.2	12100. 12100. 13148. 0.9663	602. 602. 654. 6.76 0.0481	-192. -192. -209. -0.0153	11446. 11446. 12431. 0.1739	593. 593. 644. 0.0045	2430. 2430. 2641. 0.0060	563.5 738. 0.301	59503. 53362. 1009.6 1096.8 0.007225
7	168.6 87.1 0.243 99. 685.0	0.0 0.0 0.2153 0.9045	0.0 50.3 49.4 49.1 32.5	15295. 15295. 16409. 0.9705	1019. 1019. 1127. 11.76 0.0647	-254. -254. -281. -0.0161	17186. 17186. 18999. 0.2075	2564. 2564. 2835. 0.0051	1614. 1614. 1785. 0.0032	565.7 738. 0.366	72705. 72447. 1303.3 1440.8 0.019482
8	150.5 69.1 0.219 100. 689.7	44.0 0.0 0.2141 0.9066	30.0 49.8 74.6 50.0 36.9	11680. 11680. 12969. 0.9339	472. 472. 524. 6.43 0.0377	-252. -252. -280. -0.0201	3440. 3440. 4410. 0.0607	176. 176. 195. 0.0004	-1060. -1060. -1186. -0.0027	566.7 742. 0.343	82193. 79751. 1456.2 1616.9 0.010476
9	168.9 85.8 0.248 104. 692.2	44.0 0.0 0.2115 0.8854	30.0 50.0 74.0 50.0 43.7	14785. 14785. 16633. 0.9515	1302. 1302. 1465. 15.17 0.0838	-307. -307. -345. -0.0196	1085. 1085. 1221. 0.0133	1500. 1500. 1688. 0.0030	-121. -121. -136. -0.0002	565.2 740. 0.385	93259. 93148. 1672.1 1881.2 0.012284
10	150.1 73.3 0.226 110. 693.4	44.0 0.0 0.2119 0.8894	30.0 50.0 74.0 50.0 35.7	12550. 12550. 14116. 0.9957	807. 807. 907. 11.66 0.0640	-325. -325. -365. -0.0245	-226. -226. -254. -0.0032	247. 247. 278. 0.0006	-1078. -1078. -1065. -0.0039	564.6 734. 0.356	82433. 79730. 1444.5 1624.1 0.010627
11	150.5 73.6 0.227 111. 694.0	45.0 0.0 0.2116 0.8878	45.0 50.2 74.0 50.0 42.9	11370. 11370. 12465. 0.8517	1014. 1014. 1142. 13.78 0.0761	-276. -276. -310. -0.0209	6223. 6223. 7010. 0.0489	-503. -503. -567. -0.0012	-1089. -1089. -1226. -0.0025	565.2 740. 0.357	92180. 86640. 1603.7 1806.3 0.011745
12	150.0 73.2 0.226 111. 694.0	46.0 0.0 0.2116 0.8879	52.0 50.2 74.1 50.0 42.9	11925. 11925. 13430. 0.8445	1770. 1770. 1993. 24.26 0.1335	-164. -164. -185. -0.0124	6517. 6517. 7479. 0.0493	-1332. -1332. -1500. [illegible]	-3511. -3511. -3924. [illegible]	565.4 740. 0.356	83459. 85725. [illegible] 1865.3 0.011834
13	152.0 52.9 0.191 111. 694.0	46.0 0.0 0.2126 0.8944	50. 50.0 74.0 50.0 33.7	9113. 9113. 10266. 0.9532	576. 576. 637. 10.79 0.0546	-143. -143. -160. -0.0157	2498. 2496. 2774. [illegible]	-1456. -1456. -2165. [illegible]	-2476. -2476. -2771. -0.0081	564.9 739. 0.301	74662. 68439. 1207.1 1416.2 0.004264
14	152.1 52.8 0.191 111. 694.0	46.0 0.0 0.2120 0.8944	50.0 50.3 74.4 50.0 41.6	10555. 10555. 11902. 1.1030	544. 544. 607. 11.30 0.0569	-103. -105. -119. -0.0173	1176. 1176. 1311. 0.0235	-1200. -1200. -1344. -0.0039	-1444. -1449. [illegible] -0.0065	555.3 744. 0.351	85044. 83810. 1556.4 1733.5 0.011334
15	152.1 52.0 0.191 110. 693.4	47.0 0.0 0.2224 0.2959	60. 50.3 74.0 50.0 41.5	9520. 9520. 10676. 0.9935	502. 502. 500. 5.40 0.0524	-102. -102. -114. [illegible]	2551. 2551. 2847. 0.0517	-1024. -1024. -1513. -0.0031	-2557. -2557. -2855. -0.0065	565.3 740. 0.301	88806. 84003. 1556.0 1750.1 0.011242

ORIGINAL PAGE IS
OF POOR QUALITY

TABLE V.— Continued.

RUN 19 TEST 525

PT	VELS / QPSF / MACH / TEMP / VSNDKTS	ALFS,C / PST / RHO,ICU / RHOR	INA / PED POS / LONG STK / LAT STK / PWR LEV	LIFT,C / LIFT / LIFT/W / CL	DRAG,C / DRAG / DRAG/W / FE / CD	SIDE,C / SIDE / SIDE/W / CY	PITCH,C / PITCH / PITCH/W / CM	YAW,C / YAW / YAW/W / CN	ROLL,C / ROLL / ROLL/W / CROLL	RPM / OMEGAR / V/OR	THRST,C / THRST,W / HP / HP/W / CP/SIGMA
16	150.3	+7.0	80.0	15775.	2009.	+200.	6781.	+944.	+2984.	564.9	40082.
	68.0	0.0	50.1	15775.	2009.	+200.	6781.	+944.	+2984.	734.	44279.
	0.210		74.1	12113.	2257.	+220.	7025.	+725.	+3332.	0.349	1900.3
	111.	0.2214	50.0		24.5						1763.0
	694.0	0.8895	41.3	0.2745	0.1630	+0.0162	0.1043	+0.0116	+0.0075		0.01524
17	79.4	+7.0	80.0	3610.	201.	+38.	5044.	+73.	+1140.	565.5	26442.
	19.3	0.0	50.0	3570.	201.	+38.	5044.	+73.	+1140.	740.	27449.
	0.115		56.6	3485.	222.	+42.	6227.	+807.	+1198.	0.161	509.9
	110.	0.2159	50.0		10.40						502.5
	693.4	0.9064	17.2	1.0317	0.0574	+0.0104	1.3070	+0.0065	+0.0042		0.003066

RUN 20 TEST 525

PT	VELS / QPSF / MACH / TEMP / VSNDKTS	ALFS,C / PST / RHO,ICU / RHOR	INA / PED POS / LONG STK / LAT STK / PWR LEV	LIFT,C / LIFT / LIFT/W / CL	DRAG,C / DRAG / DRAG/W / FE / CD	SIDE,C / SIDE / SIDE/W / CY	PITCH,C / PITCH / PITCH/W / CM	YAW,C / YAW / YAW/W / CN	ROLL,C / ROLL / ROLL/W / CROLL	RPM / OMEGAR / V/OR	THRST,C / THRST,W / HP / HP/W / CP/SIGMA
8	78.5	+8.0	90.0	6700.	1140.	+83.	7053.	+115.	-387.	565.8	26030.
	20.1	0.0	49.8	6700.	1140.	+83.	7053.	+115.	-387.	734.	25432.
	0.117		55.0	6953.	1183.	+86.	7320.	+119.	-402.	0.174	434.9
	75.	0.2240	50.0		55.78						477.3
	671.7	0.9635	11.7	1.9437	0.3137	+0.0228	1.3695	+0.0010	+0.0035		0.00541
9	76.8	-8.0	90.0	13570.	134.	+119.	+1299.	-756.	-2441.	563.4	67942.
	20.2	0.0	49.7	13570.	134.	+119.	+1299.	-756.	-2441.	734.	65501.
	0.117		60.5	14422.	139.	+124.	+1350.	-786.	-2537.	0.164	1191.3
	76.	0.2250	49.7		0.53						1238.2
	672.4	0.9617	23.4	3.7546	0.0366	+0.0325	+0.0075	+0.0064	+0.0250		0.01614
10	78.9	-8.0	90.0	14420.	+36.	+109.	+3848.	-515.	-3311.	565.4	75482.
	20.2	0.0	49.7	14420.	+35.	+109.	+3848.	-515.	-3311.	738.	72529.
	0.117		72.2	15030.	+38.	+114.	+4016.	-534.	-3455.	0.180	1324.4
	78.	0.2277	50.2		+1.75						1382.2
	673.6	0.9581	24.1	3.9992	+0.0099	+0.0299	+0.2805	+0.0004	-0.0282		0.01904
11	78.8	-6.0	90.0	13420.	360.	+96.	1913.	-314.	-2604.	564.6	60147.
	20.1	0.0	50.5	13420.	360.	+96.	1913.	-314.	-2604.	738.	59120.
	0.117		64.0	14084.	378.	+103.	2008.	-329.	-2467.	0.166	1067.2
	81.	0.2265	50.1		17.95						1120.0
	675.5	0.9529	20.2	3.6970	0.0992	+0.0270	+0.1003	+0.0027	-0.0241		0.01962
12	91.1	-8.0	90.0	15580.	477.	+84.	+4556.	+491.	+2035.	554.0	92300.
	26.6	0.0	51.1	15580.	477.	+84.	+4556.	+491.	+2035.	738.	93569.
	0.135		73.4	16416.	503.	+89.	+4801.	+517.	+2144.	0.200	1564.0
	82.	0.2256	50.3		17.41						1733.6
	676.1	0.9490	37.3	3.2316	0.0989	+0.0174	+0.1798	+0.0032	+0.0131		0.01527
13	79.7	-6.0	90.0	15610.	-396.	+8.	+9111.	-5184.	-658.	564.0	99152.
	20.6	0.0	40.7	15610.	-396.	+8.	+9111.	-5184.	-658.	738.	100936.
	0.118		78.0	16464.	-418.	+8.	+9610.	+5468.	-694.	0.182	1759.7
	84.	0.2254	50.7		-19.40						1857.6
	677.4	0.9481	41.2	4.2248	+0.1072	+0.0022	+0.4692	-0.0436	+0.0355		0.02405
14	90.0	+8.0	90.0	14885.	266.	+52.	-7990.	+4387.	-1376.	554.2	84533.
	26.1	0.0	40.7	14885.	266.	+52.	-7990.	-4387.	-1376.	738.	85804.
	0.134		78.9	15798.	282.	+55.	+8475.	+4653.	-1460.	0.207	1523.7
	86.	0.2241	50.6		10.20						1606.[illegible]
	678.6	0.9428	34.4	3.1524	0.0563	+0.0110	+0.3220	-0.0249	+0.0104		0.01601
15	96.2	+8.1	90.0	15655.	429.	2.	+10135.	+4448.	-2249.	555.9	98471.
	29.5	0.0	40.7	15655.	429.	2.	+10135.	+4448.	-2249.	738.	100[illegible]
	0.142		81.6	16624.	456.	2.	+10762.	+4723.	-2504.	0.220	1741.3
	86.	0.2236	50.6		14.55						1842.0
	678.6	0.9417	41.9	2.9330	0.0804	0.0003	+0.3813	-0.0259	+0.0131		0.02441
16	79.6	-12.0	90.0	10945.	-420.	+4.	610.	-3849.	235.	564.1	63597.
	20.2	0.0	40.6	10945.	-420.	+4.	610.	-3849.	235.	738.	62105.
	0.117		69.9	11607.	-445.	+4.	605.	+4082.	252.	0.182	1125.6
	87.	0.2241	50.6		-20.77						1193.8
	679.2	0.9433	23.0	2.9909	+0.1140	+0.0011	0.0424	+0.0327	0.0022		0.00784
17	79.8	-10.0	90.0	12010.	+149.	+50.	+733.	+4041.	1350.	564.4	62290.
	20.3	0.0	40.6	12010.	+149.	+50.	-733.	-4041.	1350.	738.	51051.
	0.118		70.0	12780.	+158.	+53.	+778.	+4293.	1434.	0.182	1115.6
	88.	0.2237	50.5		+7.35						1185.2
	679.9	0.9412	23.1	3.2736	+0.0406	+0.0136	+0.0380	+0.0342	0.0114		0.00778
18	79.6	+8.0	90.0	13225.	133.	+41.	+2490.	+4410.	266.	564.0	62574.
	20.2	0.0	40.6	13225.	133.	+41.	+2490.	+4410.	266.	738.	62057.
	0.117		69.4	14052.	139.	+44.	+2646.	+4685.	304.	0.182	1115.5
	88.	0.2237	50.6		6.49						1185.2
	679.9	0.9413	23.1	3.6219	0.0359	+0.0112	+0.1296	+0.0375	0.0024		0.00788
19	79.6	+8.0	90.0	14335.	473.	+27.	+3647.	+4516.	581.	564.4	[illegible]
	20.3	0.0	40.7	14335.	473.	+27.	+3647.	+4516.	581.	734.	[illegible]
	0.118		70.0	15230.	503.	+29.	+3874.	+4585.	596.	0.182	[illegible]
	88.	0.2237	50.6		25.31						[illegible]
	679.9	0.9412	23.0	3.9027	0.1286	[illegible]	+0.1009	+0.0365	[illegible]		[illegible]
20	79.8	+4.0	90.0	15460.	845.	+64.	-5170.	-4098.	-254.	563.4	60441.
	20.3	0.0	40.6	15460.	845.	+64.	-5170.	-4098.	-254.	734.	60814.
	0.116		70.1	16425.	895.	+73.	-5499.	-4355.	-269.	0.182	1088.5
	88.	0.2237	50.6		41.54						1156.4
	679.4	0.9412	23.1	4.2190	0.2245	[illegible]	[illegible]	[illegible]	[illegible]		[illegible]
21	79.9	-8.0	90.0	[illegible]	172.	+613.	-582.	-1616.	+1574.	564.[illegible]	[illegible]
	20.3	[illegible]	40.6	[illegible]	172.	+615.	-582.	-1616.	-1574.	734.	[illegible]
	0.118		70.0	13875.	183.	+653.	+614.	-1720.	-1645.	[illegible]	[illegible]
	89.	0.2237	50.6		[illegible]						1165.6
	680.5	0.9342	23.1	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]		[illegible]
22	79.2	-8.0	[illegible]	12985.	151.	+550.	+1414.	+2731.	-1503.	564.[illegible]	[illegible]
	20.2	-2.0	[illegible]	12985.	157.	+550.	+1414.	-2731.	-1503.	734.	[illegible]
	0.117		[illegible]	13654.	167.	+573.	+1504.	+2915.	+1603.	0.182	1084.6
	90.	[illegible]	50.0		[illegible]						1150.4
	680.1	0.9375	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]		[illegible]
23	79.9	[illegible]	90.0	[illegible]	142.	225.	+1850.	+5239.	1594.	[illegible]	[illegible]
	20.2	[illegible]	[illegible]	[illegible]	142.	225.	+1850.	+5239.	[illegible]	734.	[illegible]
	0.117		[illegible]	13949.	151.	[illegible]	+1952.	[illegible]	[illegible]	[illegible]	[illegible]
	[illegible]	[illegible]	[illegible]		[illegible]						[illegible]
	[illegible]	0.9375	[illegible]	3.5566	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]		[illegible]

TABLE V.— Continued.

RUN 28 TEST 525

PT	VELS / QPSF / MIUN / TEMP / VSNORTS	ALPHA / PSI / — / HORIZ / HHOF	RESTRCS / LONG STA / LAT STA / POW LEV	LIFT / LIFT / LIFT/Q / CL	DRAG / DRAG / DRAG/Q / FE / CD	SIDE / SIDE / SIDE/Q / CY	PITCH / PITCH / PITCH/Q / CM	YAW / YAW / YAW/Q / CN	ROLL / ROLL / ROLL/Q / CROLL	RPM / OMEGA / V/OR	MAST / MAST / HP / HP/Q / C/SIGMA
24	79.9	-8.0	90.0	13056.	152.	536.	-6561.	-6390.	3314.	564.2	61041.
	20.2	0.0	40.8	13056.	152.	536.	-6560.	-6390.	3314.	739.	60311.
	0.117		70.0	13445.	162.	573.	-6594.	-6664.	3542.	0.183	1066.3
	91.	0.2224	50.6		7.52						1160.8
	681.7	0.9358	23.0	3.5655	0.0415	0.1464	-0.0338	-0.0343	0.0241		0.007621
25	79.8	-8.0	90.0	12440.	167.	511.	-6625.	-6248.	3490.	564.4	60421.
	20.2	0.0	40.6	12440.	167.	511.	-6625.	-6248.	3490.	739.	60078.
	0.117		70.0	14627.	176.	546.	-6862.	-6730.	3724.	0.182	1063.6
	91.	0.2224	50.6		8.22						1157.9
	681.7	0.9358	23.1	3.5439	0.0457	0.1395	-0.0436	-0.0536	0.0297		0.007592
26	79.8	-8.0	90.0	12745.	262.	1105.	3665.	-9245.	5091.	564.2	60561.
	20.1	8.0	40.5	12745.	262.	1105.	3665.	-9245.	5091.	739.	59491.
	0.117		70.1	13643.	264.	1163.	3925.	-9450.	5449.	0.182	1074.9
	92.	0.2220	50.6		15.03						1150.7
	682.3	0.9341	23.1	3.4906	0.0718	0.3030	0.1912	-0.0792	0.0434		0.007553
27	79.5	-8.0	90.0	12390.	363.	1556.	6437.	-11372.	7295.	564.0	60341.
	25.0	12.0	40.6	12390.	363.	1556.	6437.	-11372.	7295.	739.	58964.
	0.117		70.0	13203.	410.	1660.	7031.	-12173.	7604.	0.182	1066.7
	92.	0.2220	50.6		19.15						1146.0
	682.1	0.9342	23.0	3.4218	0.1058	0.4297	0.0438	-0.0976	0.0626		0.007495
28	79.6	-8.0	90.0	12260.	533.	3127.	13126.	-9522.	-18949.	564.2	60641.
	25.0	16.0	40.7	12260.	533.	3127.	13126.	-9522.	-18949.	739.	59315.
	0.117		70.0	13054.	571.	3347.	14055.	-10193.	-20337.	0.182	1073.4
	92.	0.2220	50.6		26.61						1149.5
	682.3	0.9342	23.1	3.3653	0.1470	0.8626	0.0691	-0.0416	-0.1624		0.007546
29	79.7	-8.0	90.0	13675.	606.	-3.	6044.	-4686.	356.	564.	60841.
	20.1	0.0	40.8	13675.	606.	-3.	6044.	-4686.	356.	736.	59042.
	0.117		59.2	14665.	652.	-3.	6541.	-4375.	354.	0.182	1072.9
	93.	0.2215	49.8		30.29						1150.6
	683.3	0.9325	23.0	3.7031	0.1673	-0.0008	0.3144	-0.0349	0.0305		0.007559
30	80.0	-8.0	90.0	12310.	-156.	-70.	-7662.	-4037.	-1007.	564.3	60801.
	20.2	0.0	40.8	12310.	-156.	-70.	-7662.	-4037.	-1007.	739.	58886.
	0.117		74.2	13202.	-164.	-75.	-8453.	-4329.	-1080.	0.183	1069.8
	93.	0.2216	49.9		-7.81						1147.3
	683.0	0.9324	23.1	3.3633	-0.0432	-0.0191	-0.4648	-0.0343	-0.0066		0.007524
31	79.8	-8.0	90.0	12475.	163.	-96.	-2617.	-5303.	17431.	564.1	59295.
	20.1	0.0	40.7	12475.	163.	-96.	-2617.	-5303.	17431.	736.	57962.
	0.117		70.1	13376.	175.	-103.	-2866.	-5667.	18694.	0.183	1056.4
	93.	0.2216	64.8		8.10						1135.
	683.0	0.9325	23.1	3.4247	0.0447	-0.0264	-0.1367	-0.0453	0.1466		0.007456
32	79.7	-8.0	90.0	11760.	266.	-71.	-2776.	-5135.	37765.	564.3	60240.
	20.1	0.0	40.8	11760.	266.	-71.	-2776.	-5135.	37765.	739.	75671.
	0.117		70.2	12611.	307.	-76.	-2979.	-5507.	40521.	0.182	1057.9
	93.	0.2216	69.3		14.25						1113.1
	683.0	0.9325	23.1	3.2361	0.0787	-0.0195	-0.1455	-0.0439	0.3232		0.007303
33	79.9	-8.0	90.0	12730.	157.	-170.	-1706.	3293.	-7.24.	564.0	60002.
	20.2	0.0	60.3	12730.	157.	-170.	-1706.	3293.	-7.24.	739.	57539.
	0.117		70.2	13652.	168.	-182.	-1830.	3532.	-7840.	0.183	1052.9
	93.	0.2216	50.2		7.78						1124.2
	683.0	0.9325	23.1	3.4664	0.0430	-0.0466	-0.0869	0.0280	-0.0606		0.007396
34	79.7	-8.0	90.0	12690.	199.	-275.	-1356.	10695.	-14761.	564.0	59742.
	20.1	0.0	80.1	12690.	199.	-275.	-1356.	10695.	-14761.	739.	57695.
	0.117		70.2	13555.	213.	-295.	-1456.	11464.	-15857.	0.182	1052.0
	93.	0.2216	50.1		9.92						1128.1
	683.0	0.9325	23.0	3.4825	0.0548	-0.0758	-0.0712	0.0916	-0.1266		0.007392
35	94.5	-8.0	90.0	12015.	334.	-46.	-12022.	-3393.	-1706.	564.3	64117.
	22.7	0.0	40.7	12015.	334.	-48.	-12022.	-3393.	-1706.	734.	60838.
	0.139		69.2	12444.	360.	-52.	-12451.	-3651.	-1836.	0.216	1110.9
	94.	0.2206	50.3		11.84						1203.3
	683.6	0.9282	25.9	2.3639	0.0657	-0.0094	-0.4500	-0.0267	-0.0104		0.007907
36	100.9	-10.0	90.0	12040.	317.	-82.	-9377.	-3555.	-3290.	564.0	75741.
	31.9	0.0	40.5	12040.	317.	-82.	-9377.	-3555.	-3290.	734.	72119.
	0.146		69.4	13035.	343.	-84.	-10063.	-3840.	-3562.	0.230	1324.5
	96.	0.2196	50.6		9.95						1434.0
	684.8	0.9237	30.6	2.0880	0.0550	-0.0142	-0.3263	-0.0192	-0.0177		0.009394
37	109.9	-12.0	90.0	11925.	416.	156.	-7335.	-7436.	-942.	564.3	95596.
	37.6	0.0	31.7	11925.	416.	156.	-7335.	-7436.	-942.	739.	91617.
	0.161		91.7	12459.	454.	172.	-7977.	-8081.	-915.	0.251	1616.3
	97.	0.2187	50.6		11.12						1621.7
	685.3	0.9207	42.7	1.7523	0.0614	0.0232	-0.2051	-0.0340	-0.0038		0.01195.
38	122.2	-12.0	90.0	9610.	1654.	147.	-5346.	-7685.	-1149.	564.7	75022.
	36.2	0.0	31.6	9610.	1654.	147.	-5346.	-7455.	-1149.	739.	69794.
	0.179		91.7	10513.	1809.	161.	-5844.	-8034.	-1257.	0.279	1297.7
	99.	0.2173	50.4		35.77						1019.5
	686.7	0.9181	30.6	1.1982	0.1976	0.0176	-0.1216	-0.0293	-0.0043		0.009292
39	122.4	-12.0	90.0	11065.	1450.	170.	-3889.	-6736.	160622.	564.6	66947.
	46.3	0.0	31.7	11065.	1450.	170.	-3889.	-6736.	160622.	739.	63205.
	0.179		91.7	12105.	1586.	186.	-4254.	-7556.	175722.	0.279	1524.3
	99.	0.2173	50.5		31.36						1667.5
	686.7	0.9181	39.5	1.3195	0.1729	0.0203	-0.0842	-0.0324	0.5645		0.010925
40	122.5	-12.0	90.0	11685.	1390.	144.	-5432.	-4237.	-734.	564.2	45430.
	46.3	0.0	31.7	11685.	1390.	144.	-5432.	-4237.	-734.	734.	41227.
	0.179		91.7	12696.	1525.	217.	-5955.	-10125.	-805.	0.280	1671.0
	130.	0.2169	50.5		30.66						165[illegible].3
	687.3	0.9125	44.1	1.3934	0.1658	0.0216	-0.0233	-0.0342	-0.0027		0.012022
41	134.3	-8.0	75.1	11035.	1634.	35.	-6429.	-6401.	-3437.	564.4	46415.
	59.6	0.0	46.6	11035.	1634.	35.	-6429.	-6467.	-3437.	734.	40017.
	0.201		91.6	12076.	1814.	38.	-7004.	-7712.	-3604.	0.318	1669.4
	103.	0.2147	50.6		27.40						1647.0
	689.1	0.9034	44.5	1.0425	0.1525	0.0033	-0.1226	-0.0421	-0.0349		0.012126
42	139.3	-8.0	75.0	10303.	2124.	31.	-5791.	-6561.	-3770.	564.7	60937.
	54.3	0.0	40.5	10365.	2124.	31.	-5791.	-6507.	-3760.	734.	73694.
	0.201		91.7	11443.	2355.	34.	-6427.	-7262.	-4191.	0.338	1360.3
	104.	0.2144	50.5		35.53						1537.3
	689.7	0.9014	35.2	0.9001	0.1946	0.0024	-0.1026	-0.0114	-0.0116		0.010666

TABLE V.— Continued.

RUN 20 TEST 525

PT	VKTS QPSF MTUN TEMP VSNDKTS	ALFS,C PSI RHOSTD RHOR	INA PED POS LONG STK LAT STK PCH LEV	LIFT,C LIFT LIFT/W CL	DRAG,C DRAG DRAG/W FE CD	SIDE,C SIDE SIDE/W CY	PITCH,C PITCH PITCH/W CM	YAW,C YAW YAW/W CN	ROLL,C ROLL ROLL/W CROLL	RPM RPMEAN V/TIP	WHAST,C WHAST HP HP/W CP/SIGMA
43	139.7 59.5 0.203 105. 690.3	+8.0 0.0 0.2146 0.9002	75.0 40.0 41.0 50.5 30.5	5595. 5595. 10859. 0.8913	2058. 2058. 2730. 41.33 0.2243	31. 31. 34. 0.0024	-4556. -4506. -4450. +0.0772	+6031. +6031. +7590. +0.0047	+5126. +5126. +5644. +0.0148	564.5 734. 0.314	12100. 8452. 1220.5 1362.2 0.000924
44	139.5 59.5 0.203 105. 693.3	+8.0 0.0 0.2140 0.9002	75.0 40.6 41.4 50.4 25.6	8845. 8845. 9825. 0.8207	2819. 2819. 3132. 47.34 0.2615	25. 25. 26. 0.0023	-3534. -3534. -3420. +0.0024	+6540. +6540. -7050. -0.0163	+5911. +5911. +4214. +0.0110	564.7 734. 0.315	81740. 5496. 1045.7 1501.7 0.000706
45	100.6 77.7 0.233 109. 692.2	+8.0 0.0 0.2115 0.8697	45.0 40.5 65.4 50.5 50.6	10910. 10910. 12263. 0.7759	756. 756. 850. 9.73 0.0538	83. 83. 93. 0.0059	4563. 4563. 5129. 0.0018	-8054. -8054. -9053. +0.0110	+1950. +1950. +2113. +0.0042	564.6 734. 0.307	115045. 110441. 2054.0 2309.5 0.000325
46	145.0 63.5 0.210 109. 692.2	+5.0 0.0 0.2325 0.8981	45.0 40.6 84.1 50.3 7.6	6195. 6095. 6817. 0.5203	4846. 4846. 5420. 76.19 0.4209	-17. -17. -19. +0.0015	-24584. -24584. -27271. +0.4030	-7327. -7327. -8144. +0.0194	+3542. -3542. -4617. +0.0097	562.7 737. 0.332	83.4 -3125. -4.3 51.9 0.000343

RUN 21 TEST 525

PT	VKTS QPSF MTUN TEMP VSNDKTS	ALFS,C PSI RHOSTD RHOR	INA PED POS LONG STK LAT STK PCH LEV	LIFT,C LIFT LIFT/W CL	DRAG,C DRAG DRAG/W FE CD	SIDE,C SIDE SIDE/W CY	PITCH,C PITCH PITCH/W CM	YAW,C YAW YAW/W CN	ROLL,C ROLL ROLL/W CROLL	RPM RPMEAN V/TIP	WHAST,C WHAST HP HP/W CP/SIGMA
3	81.0 20.6 0.118 97. 685.4	-8.0 0.0 0.2198 0.9237	60.0 43.8 63.0 50.1 20.3	5360. 5360. 5747. 1.4905	-1154. -1154. -1245. -56.13 -0.3101	-5. -5. -5. -0.0013	3717. 3717. 4014. 0.1901	-1251. -1251. -1352. [illegible]	393. 393. 425. 0.0033	564.5 734. 0.185	49376. 50025. 840.4 902.4 0.000310
4	81.1 20.5 0.116 98. 655.0	-4.0 0.0 0.2194 0.9230	60.0 43.6 63.0 50.1 20.2	7005. 7005. 7584. 1.8648	+991. +991. +1074. -48.26 -0.2666	-3. -3. -3. -0.0006	1962. 1962. 2126. 0.1005	+1752. -1752. -1898. -0.0147	-267. -267. -289. -0.0022	564.5 734. 0.185	49815. 48775. 880.4 954.0 0.000202
5	80.9 20.4 0.118 99. 686.7	+2.0 0.7 0.2190 0.9214	60.0 43.8 63.0 50.2 20.0	7805. 7805. 8471. 2.1059	-877. -877. -952. -42.91 -0.2371	4. 4. 4. 0.0011	-32. -32. -34. -0.0016	+2110. -2110. -2289. +0.0177	+86. +86. +94. +0.0007	564.7 734. 0.185	49549. 48570. 880.7 955.0 0.000254
6	80.9 20.6 0.119 100. 687.3	0.0 0.0 0.2188 0.9198	60.0 43.7 63.1 50.0 20.0	8460. 8460. 9190. 2.2924	-729. -729. -793. -35.75 -0.1975	-14. -14. -15. -0.0036	+1232. +1232. +1339. +0.0630	-1750. -1750. -1903. +0.0147	410. 410. 440. 0.0035	564.4 734. 0.185	48557. 46307. 867.5 943.1 0.000184
7	81.1 20.5 0.116 100. 687.3	2.0 0.0 0.2186 0.9198	60.0 43.6 63.0 50.1 20.0	9245. 9245. 10051. 2.4962	-614. -614. -668. -30.01 -0.1658	2. 2. 2. 0.0005	+1329. +1329. +1445. +0.0683	-1697. -1697. -2005. +0.0159	85. 85. 94. 0.0007	564.1 734. 0.185	48557. 48284. 866.7 942.4 0.000141
8	81.4 20.7 0.119 100. 687.3	4.0 0.0 0.2186 0.9197	60.0 43.8 63.1 50.0 20.3	9960. 9960. 10830. 2.6845	-457. -457. -497. -22.13 -0.1222	-8. -8. -9. -0.0021	+2575. +2575. +2800. +0.1362	-1745. -1745. -1899. +0.0145	780. 780. 849. 0.0065	564.5 734. 0.186	48457. 47619. 862.2 937.5 0.000147
9	80.8 20.3 0.118 101. 687.9	6.0 0.0 0.2182 0.9182	60.0 43.4 63.0 50.2 20.0	10630. 10630. 11577. 2.8974	-361. -361. -393. -17.81 -0.0984	8. 8. 9. 0.0022	+3304. +3309. +3604. +0.1718	-2064. -2064. -2248. +0.0175	284. 284. 310. 0.0024	564.5 734. 0.185	48572. 47024. 859.8 936.2 0.000142
10	81.2 20.4 0.118 102. 688.5	7.0 0.0 0.2176 0.9165	60.0 43.6 63.1 50.2 20.0	11665. 11665. 12675. 2.9472	-250. -250. -273. -12.25 -0.0676	1. 1. 1. 0.0003	+3915. +3915. +4271. +0.2014	-1777. -1777. -1935. +0.0149	-12. -12. -14. +0.0000	564.5 734. 0.185	47444. 47242. 855.5 931.5 0.000104
11	120.2 44.2 0.175 106. 695.9	+5.0 0.0 0.2145 0.9026	60.0 39.6 64.0 50.1 24.0	9275. 9275. 10276. 1.1004	787. 787. 872. 17.82 0.1965	82. 82. 94. 0.0072	7514. 7514. 8321. 0.1768	+3002. -3082. -4501. -0.0151	1241. 1241. 1375. 0.0064	564.7 734. 0.274	80720. 80010. 1342.2 1494.1 0.000564
12	119.7 43.8 0.174 106. 690.9	-5.0 0.0 0.2146 0.9027	60.0 40.0 64.0 50.3 24.5	10670. 10670. 11820. 1.3484	427. 427. [illegible] [illegible] [illegible]	25. 25. 26. 0.0032	4647. 4647. 5144. [illegible]	+4365. +4365. +4810. [illegible]	148. 148. 164. [illegible]	564.5 734. 0.275	[illegible] 57626. 1082.4 [illegible] [illegible]
13	119.8 43.8 0.174 107. 691.5	+1.0 0.0 0.2142 0.9010	60.0 40.0 64.0 50.4 24.5	12095. 12095. 13423. 1.5251	1126. 1126. 1243. 25.50 0.1412	21. 21. 23. 0.0026	2404. 2404. 2674. [illegible]	+4675. -4675. -5102. [illegible]	-681. -681. -756. [illegible]	564.5 734. 0.274	60222. 57645. 1096.1 [illegible] 0.000633
14	120.2 44.0 0.174 107. 691.5	1.0 0.0 0.2142 0.9010	60.0 40.0 64.0 50.0 24.0	13570. 13570. 15051. 1.7015	1304. 1304. 1441. 29.59 0.1635	22. 22. 24. 0.0020	445. 445. 491. [illegible]	-4767. -4767. -5290. [illegible]	-130. -130. -144. [illegible]	564.2 734. 0.275	60142. 58285. 1115.0 1275.1 0.000715

TABLE V.— Continued.

RUN 21 TEST 525

PT	VKTS / QPSF / RTOR / TEMP / VSNDKTS	ALFS,C / PSI / RHO*10^6 / RHOR	INA / PED POS / LONG STK / LAT STK / PWR LEV	LIFT,U / LIFT / LIFT/Q / CL	DRAG,U / DRAG / DRAG/Q / FE / CD	SIDE,U / SIDE / SIDE/Q / CY	PITCH,U / PITCH / PITCH/Q / CM	YAW,U / YAW / YAW/Q / CN	ROLL,U / ROLL / ROLL/Q / CROLL	RPM / OMEGAR / V/OR	HPAST G / CHAST G / HP / HP/Q / CP/SIGMA
15	120.2	2.0	60.0	14195.	1406.	+7.	-335.	-3825.	-567.	564.1	59822.
	48.0	0.0	40.0	14195.	1406.	+7.	-335.	-3825.	-567.	736.	58598.
	0.174		64.7	15771.	1563.	+8.	-373.	-4250.	-563.	0.275	1058.2
	108.	0.2238	50.2		31.44						1176.5
	692.2	0.8993	20.0	1.7822	0.1765	+0.0009	-0.0080	-0.0944	-0.0127		0.007720
16	120.1	3.0	60.0	14925.	1475.	-26.	-1644.	-3864.	-149.	564.1	59363.
	43.8	0.0	40.0	14925.	1475.	-26.	-1644.	-3869.	-149.	736.	58417.
	0.174		64.7	16594.	1640.	-29.	-1828.	-4230.	-166.	0.274	1054.2
	108.	0.2197	50.0		33.49						1172.5
	692.2	0.8992	22.5	1.8741	0.1856	-0.0033	-0.0393	-0.0149	-0.0066		0.007703

RUN 22 TEST 525

PT	VKTS / QPSF / RTOR / TEMP / VSNDKTS	ALFS,C / PSI / RHO*10^6 / RHOR	INA / PED POS / LONG STK / LAT STK / PWR LEV	LIFT,U / LIFT / LIFT/Q / CL	DRAG,U / DRAG / DRAG/Q / FE / CD	SIDE,U / SIDE / SIDE/Q / CY	PITCH,U / PITCH / PITCH/Q / CM	YAW,U / YAW / YAW/Q / CN	ROLL,U / ROLL / ROLL/Q / CROLL	RPM / OMEGAR / V/OR	HPAST G / CHAST G / HP / HP/Q / CP/SIGMA
3	46.9	0.0	40.0	12875.	843.	+24.	-5703.	-3375.	1547.	565.0	67015.
	7.2	0.0	40.0	12875.	843.	+24.	-5703.	-3375.	1547.	744.	65321.
	0.070		64.1	13659.	850.	+25.	-5673.	-3476.	1543.	0.107	827.7
	75.	0.2308	50.0		51.23						852.4
	672.7	0.9710	15.8	4.8322	0.3383	+0.0183	-0.8287	-0.0801	0.0367		0.005574
4	60.8	0.0	40.0	12770.	716.	-16.	+7970.	-3530.	3267.	564.7	37811.
	12.1	0.0	40.0	12770.	716.	-16.	+7970.	-3530.	3267.	739.	35991.
	0.091		58.9	13143.	740.	-17.	+8234.	-3647.	3375.	0.119	661.3
	76.	0.2301	50.0		59.10						685.2
	672.0	0.9679	12.4	5.3292	0.3266	-0.0073	+0.6923	-0.0501	0.0464		0.004473
5	79.2	0.0	40.0	12965.	1426.	+20.	-7106.	+3275.	4096.	564.9	32678.
	20.75	0.0	39.9	12965.	1426.	-20.	-7106.	+3275.	4096.	739.	30604.
	0.118		73.9	13479.	1482.	+21.	+7376.	-3400.	4252.	0.181	567.6
	77.	0.2290	50.0		64.79						583.2
	673.0	0.9634	12.5	3.5061	0.3856	-0.0054	+0.3051	+0.0275	0.0354		0.003814
6	117.4	-1.0	60.0	12495.	1046.	-1237.	-7004.	2647.	-7875.	564.1	63017.
	43.3	-4.0	40.1	12495.	1046.	-1237.	-7004.	2647.	-7875.	738.	57871.
	0.175		74.1	13150.	1095.	-1302.	+7413.	2786.	+8288.	0.265	1067.3
	90.	0.2258	50.7		25.45						1144.3
	678.4	0.9502	19.8	1.5567	0.1256	-0.1541	-0.1970	0.0703	-0.0365		0.007518
7	117.7	-1.0	60.0	12370.	1034.	+649.	+6830.	760.	-2603.	563.8	61540.
	44.3	-2.0	40.1	12370.	1034.	-639.	-6800.	760.	-2603.	738.	58730.
	0.175		74.1	13044.	1094.	+687.	-7263.	804.	-2755.	0.264	1074.1
	83.	0.2246	50.0		23.34						1136.7
	676.7	0.9449	19.8	1.5427	0.1250	+0.0809	-0.1615	0.0024	-0.0101		0.007474
8	117.7	-1.0	50.0	12365.	1052.	675.	-6381.	+7482.	5184.	563.9	58324.
	44.3	-2.0	39.5	12365.	1052.	675.	-6381.	+7482.	5184.	736.	60252.
	0.175		74.1	13066.	1113.	714.	-6755.	-7916.	5467.	0.264	1063.4
	83.	0.2246	50.7		23.75						1122.6
	676.7	0.9449	19.8	1.5421	0.1312	0.0882	+0.1516	-0.0295	0.0201		0.007362
9	118.1	-1.0	60.0	12405.	1161.	1252.	+5568.	+9964.	8680.	563.6	56587.
	45.6	4.0	40.1	12405.	1161.	1252.	+5568.	+9964.	8680.	736.	60416.
	0.175		74.0	13153.	1216.	1323.	+5925.	+10544.	9231.	0.270	1049.3
	84.	0.2242	56.7		25.61						1112.8
	677.0	0.9431	19.7	1.5362	0.1415	0.1550	+0.1319	+0.0363	0.0315		0.007320
10	118.0	-1.0	60.0	12265.	1249.	+20464.	+5168.	+118004.	664595.	564.0	55391.
	44.4	0.0	40.2	12265.	1249.	+20464.	+5168.	+118004.	664595.	736.	62360.
	0.175		74.0	13028.	1327.	+22208.	+5490.	+125350.	705939.	0.270	1050.0
	85.	0.2236	51.0		28.15						1100.4
	678.0	0.9412	19.8	1.5272	0.1555	+2.6104	+0.1225	+0.4566	2.5724		0.007233
11	118.5	-1.0	60.0	12100.	1196.	21.	1263.	-4162.	624.	563.5	58124.
	44.7	0.0	40.1	12100.	1196.	21.	1263.	-4162.	624.	736.	56456.
	0.175		64.3	12678.	1275.	22.	1366.	+4430.	669.	0.271	1024.6
	86.	0.2233	50.5		26.77						1090.5
	678.6	0.9398	14.7	1.4965	0.1479	0.0026	0.0362	-0.0150	0.0024		0.007184
12	118.3	-1.0	60.0	12075.	1214.	72.	-13332.	+4275.	725.	564.1	57585.
	44.5	0.0	40.0	12075.	1214.	72.	-13332.	+4275.	725.	736.	55460.
	0.175		64.2	12650.	1242.	77.	-14186.	+4544.	772.	0.270	1015.7
	86.	0.2235	51.7		27.25						1081.0
	679.6	0.9397	19.7	1.4941	0.1516	0.0089	+0.3145	-0.0155	0.0020		0.007106
13	118.3	-1.0	60.0	12675.	1210.	10.	-7614.	2247.	19323.	563.7	41423.
	44.4	0.0	40.0	12675.	1210.	10.	-7619.	2247.	19323.	738.	60568.
	0.175		79.1	12866.	1240.	11.	+7485.	2444.	20801.	0.271	1001.0
	87.	0.2229	70.0		27.23						1067.3
	679.2	0.9360	19.7	1.5006	0.1504	0.0012	-0.1660	0.0029	0.0747		0.007027
14	115.4	-1.0	60.0	11595.	1371.	11.	+6851.	680.	43766.	563.8	27265.
	44.5	0.0	40.1	11595.	1371.	11.	+6851.	680.	43766.	738.	62405.
	0.175		74.3	12362.	1462.	10.	+7364.	917.	46641.	0.271	961.0
	87.	0.2229	[illegible]		30.62						1045.4
	679.2	0.9360	14.7	1.4400	[illegible]	0.0013	+0.1614	0.0032	0.1694		0.006892
15	118.6	-1.0	60.0	12065.	1251.	+245.	+8803.	1657.	+7444.	563.4	60302.
	44.6	0.0	60.0	12065.	1251.	+245.	+8803.	1657.	+7445.	734.	51157.
	0.175		74.1	12657.	1336.	+313.	+7055.	1985.	+7933.	0.271	997.1
	88.	0.2225	50.0		25.08						1065.0
	679.4	0.9362	19.7	1.4961	0.1551	+0.0300	+0.1550	0.0072	+0.0287		0.007006

ORIGINAL PAGE IS
OF POOR QUALITY

TABLE V.— Continued.

RUN 22 TEST 525

PT	VKTS QPSF MACH TEMP VSNDRTS	ALPS,D PSI RHO,SG RHOR	ISA PFD POS LONG STK LAT STK PDR DEV	LIFT,D LIFT LIFT/Q CL	DRAG,D DRAG DRAG/Q FE CD	SIDE,D SIDE SIDE/Q CY	PITCH,D PITCH PITCH/Q CM	YAW,D YAW YAW/Q CN	ROLL,D ROLL ROLL/Q CROLL	RPM OMEGAR V/OR	HPAST LMAST HP HP/Q CP/SIGMA
16	116.7	-1.0	80.0	12023.	1287.	-587.	-5875.	7983.	-13034.	584.1	6599.
	44.2	0.0	79.5	12023.	1287.	-587.	-5875.	7983.	-13034.	739.	44966.
	0.175		74.1	12534.	1375.	-627.	-6276.	8527.	-13786.	0.271	1013.5
	86.	0.2225	50.0		28.43						1069.
	679.9	0.9362	19.7	1.4873	0.1593	-0.0726	-0.7380	0.0307	-0.0424		0.007025
17	79.9	0.0	95.0	8200.	2090.	-3.	1724.	-2630.	1204.	457.5	8504.
	20.5	0.0	40.2	8200.	2090.	-3.	1724.	-2630.	1204.	599.	8812.
	0.118		54.9	8800.	2214.	-3.	1830.	-2788.	1275.	0.225	111.7
	88.	0.2244	50.3		102.38						116.3
	679.9	0.9391	3.5	2.2193	0.5656	-0.0008	0.0840	-0.0221	0.0101		0.001456
18	79.9	0.0	95.0	10340.	2554.	6.	6162.	-3267.	2244.	457.8	8599.
	20.5	0.0	40.5	10340.	2554.	6.	6162.	-3267.	2244.	599.	10735.
	0.118		54.8	10913.	2696.	6.	6504.	-3448.	2369.	0.225	141.1
	88.	0.2252	50.8		124.87						140.9
	678.6	0.9475	5.2	2.7286	0.6849	0.0016	[illegible]	-0.0214	0.0148		0.001836
19	80.0	2.0	95.0	10650.	2764.	12.	5546.	-3481.	2107.	457.8	5355.
	20.5	0.0	40.2	10650.	2764.	12.	5546.	-3481.	2107.	599.	6751.
	0.118		54.7	11261.	2923.	13.	5864.	-3680.	2082.	0.226	89.4
	87.	0.2245	51.6		134.77						94.5
	679.2	0.9457	3.5	2.8689	0.7446	0.0032	0.2843	-0.0201	0.0227		0.001162
20	79.8	4.0	95.0	11670.	3165.	-3.	5745.	-3597.	1895.	457.4	4554.
	20.4	0.0	40.3	11670.	3165.	-3.	5745.	-3597.	1895.	599.	7165.
	0.118		54.6	12339.	3351.	-3.	6074.	-3803.	2004.	0.225	95.8
	87.	0.2246	50.9		155.24						90.7
	679.2	0.9458	3.5	3.1584	0.8577	-0.0009	1.2959	-0.0305	0.0159		0.001117
21	79.9	6.0	95.0	12610.	3651.	7.	5542.	-3912.	2325.	457.4	3695.
	20.6	0.0	40.3	12610.	3651.	7.	5542.	-3912.	2325.	599.	7690.
	0.118		54.6	13333.	3860.	7.	6171.	-4136.	2458.	0.225	82.7
	87.	0.2244	50.8		178.64						87.5
	679.2	0.9456	3.5	3.4096	0.9870	0.0019	0.3605	-0.0324	0.0195		0.001074
22	79.9	8.0	95.0	11665.	3334.	12.	4924.	-2455.	-937.	457.9	1578.
	20.5	0.0	40.4	11665.	3334.	12.	4924.	-2455.	-937.	599.	351.
	0.118		54.6	12334.	3525.	13.	5206.	-2595.	-991.	0.225	14.0
	87.	0.2248	50.7		162.44						14.8
	679.2	0.9458	3.5	3.1447	0.9002	0.0032	0.2530	-0.0206	-0.0079		0.000182
23	80.0	8.0	95.0	12890.	3916.	0.	5081.	-2285.	-1284.	457.1	519.
	20.5	0.0	40.8	12890.	3916.	0.	5081.	-2285.	-1284.	599.	519.
	0.118		54.5	13659.	4148.	0.	5382.	-2420.	-1360.	0.226	12.0
	88.	0.2244	50.8		191.39						12.7
	679.9	0.9440	3.5	3.4834	1.0579	0.0000	0.2611	-0.0142	-0.0108		0.000157
24	80.3	8.0	95.0	10800.	3218.	15.	3163.	-2074.	-677.	452.2	-1119.
	23.6	0.0	40.4	10800.	3218.	15.	3163.	-2074.	-677.	592.	-3064.
	0.118		54.5	11444.	3409.	16.	3372.	-2197.	-717.	0.229	-30.0
	88.	0.2244	50.9		156.17						-31.8
	679.9	0.9440	3.6	2.8957	0.8628	0.0040	0.1624	-0.0175	-0.0055		-0.000405
25	80.0	10.0	95.0	13280.	4221.	16.	4538.	-1915.	-1812.	458.1	-2317.
	20.4	0.0	40.3	13280.	4221.	16.	4538.	-1915.	-1812.	600.	-2598.
	0.118		54.4	14067.	4471.	17.	4807.	-2029.	-1708.	0.225	-55.7
	88.	0.2244	50.9		236.53						-57.8
	679.9	0.9440	3.5	3.5699	1.1417	0.0045	0.2335	-0.0163	-0.0135		-0.000464
26	79.9	10.0	95.0	13152.	4164.	20.	4074.	-1775.	-1135.	457.4	-2517.
	20.4	0.0	40.2	13152.	4164.	20.	4074.	-1775.	-1135.	599.	-3201.
	0.118		54.4	13929.	4432.	21.	4321.	-1880.	-1200.	0.225	-41.5
	88.	0.2244	50.8		205.20						-44.0
	679.9	0.9441	3.5	3.5032	1.1337	0.0054	0.2163	-0.0150	-0.0045		-0.000541
27	79.8	11.0	95.0	13615.	4445.	20.	4949.	-1914.	-798.	457.2	-2858.
	20.3	0.0	40.0	13615.	4445.	20.	4949.	-1914.	-796.	598.	-2428.
	0.118		54.3	14440.	4727.	21.	5251.	-2037.	-847.	0.225	-42.0
	89.	0.2240	50.9		218.72						-44.5
	680.5	0.9429	3.4	3.7023	1.2087	0.0054	0.2561	-0.0182	-0.0067		-0.000544
28	79.5	12.0	95.0	13575.	4898.	-83.	3284.	-3163.	-2331.	457.5	-2798.
	20.1	0.0	40.0	13575.	4898.	-83.	3284.	-3163.	-2431.	599.	-2406.
	0.117		54.2	14404.	5197.	-88.	3440.	-3293.	-3008.	0.224	-41.4
	89.	0.2240	50.9		243.09						-43.9
	680.5	0.9424	3.5	3.7222	1.3432	-0.0228	0.1710	-0.0205	-0.0244		-0.000541
29	79.9	0.0	95.0	13555.	2800.	-24.	3369.	-2285.	-1586.	563.2	17557.
	20.0	0.0	40.7	13555.	2800.	-24.	3369.	-2285.	-1598.	737.	14287.
	0.118		54.2	14364.	2971.	-25.	3575.	-2423.	-1683.	0.183	284.4
	89.	0.2240	50.8		137.49						301.8
	680.5	0.9323	11.2	3.6773	0.7596	-0.0065	0.1734	-0.0193	-0.0134		0.001942
30	80.0	2.0	95.0	14190.	3175.	-14.	3259.	-2430.	-1492.	563.3	14801.
	20.4	0.0	40.1	14190.	3175.	-14.	3259.	-2430.	-1492.	737.	11744.
	0.118		54.2	15135.	3362.	-15.	3471.	-2584.	-1599.	0.183	238.0
	91.	0.2232	50.6		155.40						251.9
	681.7	0.9359	10.0	3.8495	0.8613	-0.0038	0.1683	-0.0205	-0.0126		0.001862
31	81.7	4.0	95.0	11105.	2889.	0.	2773.	-2190.	-1312.	561.6	8944.
	21.2	0.0	40.2	11105.	2889.	0.	2773.	-2190.	-1312.	735.	5797.
	0.120		54.1	11851.	3078.	0.	2954.	-2334.	-1398.	0.188	131.4
	91.	0.2231	50.6		136.60						140.1
	681.7	0.9386	9.2	2.8849	0.7518	0.0000	0.1373	-0.0178	-0.0106		0.000932
32	81.1	6.0	95.0	12230.	3298.	-13.	2466.	-2010.	-955.	563.0	8172.
	20.9	0.0	40.2	12230.	3298.	-13.	2466.	-2010.	-955.	737.	6164.
	0.119		54.0	13026.	3513.	-14.	2557.	-2141.	-1017.	0.185	94.3
	91.	0.2231	50.4		157.48						89.0
	681.7	0.9387	8.5	3.2557	0.8713	-0.0034	0.1266	-0.0165	-0.0078		0.000543

TABLE V.— Continued.

RUN 22 TEST 525

PT	VKTS / QPSF / RTUN / TEMP / VSNDRTS	ALFS,C / PSI / RHOPIDS / RROR	INA / PED POS / LONG STA / LAT STA / POS LEV	LIFT,U / LIFT / LIFT/B / CL	DRAG,U / DRAG / DRAG/B / FE / CD	SIDE,U / SIDE / SIDE/B / CY	PITCH,U / PITCH / PITCH/B / CM	YAW,U / YAW / YAW/B / CN	ROLL,U / ROLL / ROLL/B / CROLL	RPM / OMEGAR / V/OR	HHAST U / CHAST U / HP / HP/B / CP/SIGMA
33	80.1	0.0	95.0	13070.	3701.	3.	2330.	-2175.	-1455.	563.1	4175.
	20.0	0.0	40.2	13070.	3701.	1.	2330.	-2175.	-1455.	737.	436.
	0.119		54.1	13420.	3942.	3.	2462.	-2317.	-1544.	0.183	43.8
	91.	0.2232	50.9		181.30						48.6
	683.7	0.9389	6.0	3.5373	1.0017	0.0008	0.1200	-0.0093	-0.0122		0.000322
34	79.2	10.0	95.0	13675.	4110.	20.	2147.	-4067.	478.	558.3	2017.
	20.0	0.0	40.4	13675.	4110.	20.	2147.	-4067.	478.	731.	-2064.
	0.116		54.1	14562.	4377.	21.	2256.	-4331.	509.	0.183	-0.5
	91.	0.2232	50.9		205.44						-0.5
	681.7	0.9391	6.9	3.7658	1.1376	0.0055	0.1131	-0.0350	0.0411		-0.000063
35	78.2	12.0	95.0	14060.	4324.	-19.	2121.	-635.	2457.	559.8	-364.
	19.6	0.0	50.4	14060.	4324.	-19.	2121.	-635.	2457.	733.	-1971.
	0.116		54.0	14934.	4612.	-20.	2263.	-691.	2620.	0.181	-20.7
	92.	0.2226	50.8		220.12						-22.1
	682.3	0.9375	6.2	3.9375	1.2161	-0.0053	0.1135	-0.0073	0.0215		-0.000146
36	79.9	12.0	95.0	14230.	4848.	-146.	-674.	-1822.	404.	567.2	-2037.
	20.3	0.0	50.6	14230.	4848.	-142.	-674.	-1822.	404.	753.	-3445.
	0.117		54.0	15103.	5228.	-156.	-719.	-1944.	464.	0.184	-52.3
	92.	0.2225	50.4		241.64						-55.6
	682.3	0.9373	4.2	3.6727	1.3351	-0.0403	-0.0356	-0.0154	0.0077		-0.000374
37	79.9	12.0	95.0	14200.	4673.	-132.	-844.	-1967.	1031.	559.5	-2157.
	20.5	0.0	51.4	14230.	4673.	-132.	-844.	-1967.	1031.	732.	-3186.
	0.117		54.0	15151.	5199.	-141.	-900.	-2099.	1068.	0.184	-52.8
	92.	0.2228	50.9		240.41						-56.3
	682.3	0.9373	4.0	3.6765	1.3262	-0.0360	-0.0430	-0.0167	0.0365		-0.000379
38	80.4	12.0	95.0	16585.	6254.	-122.	7206.	-1956.	277.	562.2	-3635.
	20.5	0.0	50.6	16565.	6254.	-122.	7206.	-1956.	277.	736.	-2166.
	0.116		44.0	17647.	6673.	-130.	7684.	-2090.	295.	0.184	-53.5
	92.	0.2225	50.5		305.29						-57.1
	682.3	0.9372	3.4	4.4729	1.6667	-0.0324	0.3648	-0.0164	0.0023		-0.000379
39	80.5	12.0	95.0	16875.	6241.	-106.	5377.	-2941.	2745.	563.9	-2616.
	20.5	0.0	51.5	16875.	6241.	-108.	5377.	-2941.	2745.	735.	-3240.
	0.116		44.0	18006.	6659.	-115.	5737.	-3139.	2982.	0.186	-59.1
	92.	0.2224	50.8		325.01						-63.6
	682.3	0.9372	3.5	4.5514	1.6851	-0.0292	0.2763	-0.0247	0.0235		-0.000320
40	60.4	0.0	40.0	5120.	1069.	-24.	1322.	-323.	2495.	557.2	18056.
	11.7	0.0	49.5	5120.	1069.	-24.	1322.	-323.	2495.	729.	16454.
	0.069		55.1	5457.	1139.	-26.	1409.	-344.	2659.	0.143	305.1
	93.	0.2230	56.9		91.06						325.2
	683.0	0.9385	13.9	2.4097	0.5031	-0.0113	0.1164	-0.0047	0.0365		0.002216

RUN 23 TEST 525

PT	VKTS / QPSF / RTUN / TEMP / VSNDRTS	ALFS,C / PSI / RHOPIDS / RROR	INA / PED POS / LONG STA / LAT STA / POS LEV	LIFT,U / LIFT / LIFT/B / CL	DRAG,U / DRAG / DRAG/B / FE / CD	SIDE,U / SIDE / SIDE/B / CY	PITCH,U / PITCH / PITCH/B / CM	YAW,U / YAW / YAW/B / CN	ROLL,U / ROLL / ROLL/B / CROLL	RPM / OMEGAR / V/OR	HHAST U / CHAST U / HP / HP/B / CP/SIGMA
3	109.7	0.0	0.0	7130.	-3665.	-31.	-683.	-556.	-7.	565.7	102027.
	38.2	0.0	47.3	7130.	-3665.	-31.	-683.	-556.	-7.	740.	102411.
	0.162		49.9	7602.	-3907.	-33.	-729.	-593.	-8.	0.251	1835.0
	88.	0.2224	50.3		-95.93						1956.3
	675.9	0.9387	44.7	1.0310	-0.5300	-0.0045	-0.0168	-0.0025	0.0000		0.012742
4	109.7	0.0	0.0	7150.	-3681.	-118.	-1531.	960.	-366.	566.0	101728.
	38.2	0.0	50.8	7150.	-3681.	-118.	-1531.	960.	-366.	741.	102040.
	0.162		49.8	7623.	-3924.	-126.	-1633.	1024.	-390.	0.250	1830.0
	88.	0.2224	50.2		-96.29						1951.1
	679.9	0.9390	48.7	1.0333	-0.5320	-0.0171	-0.0421	0.0043	-0.0016		0.012665
5	110.4	4.0	0.0	9975.	-3360.	-96.	-3427.	567.	-86.	565.3	100489.
	38.3	0.0	51.2	9975.	-3360.	-96.	-3427.	567.	-86.	740.	99854.
	0.162		49.6	10732.	-3615.	-105.	-3667.	454.	-92.	0.252	1797.1
	93.	0.2229	50.2		-67.67						1933.5
	683.0	0.9290	44.8	1.4380	-0.4844	-0.0141	-0.0960	0.0040	-0.0000		0.012617
6	110.6	6.0	0.0	11630.	-3253.	-105.	-4960.	593.	-367.	565.3	100404.
	38.4	0.0	51.1	11630.	-3253.	-105.	-4960.	593.	-367.	740.	100205.
	0.162		44.7	12536.	-3506.	-113.	-5348.	634.	-340.	0.252	1804.0
	94.	0.2205	50.2		-64.68						1944.6
	683.6	0.9277	40.6	1.6725	-0.4676	-0.0151	-0.1357	0.0026	-0.0016		0.012688
7	110.6	8.0	0.0	13050.	-3089.	-107.	-6720.	150.	-1123.	565.5	101947.
	38.3	0.0	51.1	13050.	-3089.	-107.	-6725.	150.	-1123.	740.	99676.
	0.162		44.8	14092.	-3336.	-116.	-7256.	162.	-1213.	0.252	1809.3
	95.	0.2201	50.1		-60.55						1953.7
	684.2	0.9261	40.7	1.8802	-0.4451	-0.0154	-0.1842	0.0037	-0.0050		0.012734
8	110.6	10.0	0.0	14505.	-2850.	-143.	-6446.	318.	-1884.	555.2	100809.
	38.2	0.0	51.7	14505.	-2850.	-143.	-6446.	318.	-1869.	740.	98372.
	0.162		49.7	15662.	-3077.	-154.	-9114.	343.	-2018.	0.252	1755.0
	95.	0.2201	50.1		-74.55						1920.2
	684.2	0.9261	44.6	2.0962	-0.4119	-0.0207	-0.2321	0.0014	-0.0060		0.012596
9	110.6	12.0	0.0	15795.	-2625.	-134.	-8011.	143.	-1742.	565.5	101266.
	38.3	0.0	51.3	15795.	-2625.	-134.	-8011.	143.	-1742.	740.	98332.
	0.162		44.7	17066.	-2844.	-145.	-10013.	154.	-1885.	0.252	1741.0
	96.	0.2197	50.0		-68.56						1917.4
	684.8	0.9244	44.6	2.2794	-0.3789	-0.0193	-0.2695	0.0006	-0.0026		0.012633
10	110.4	[illegible]	0.0	12445.	-2453.	-1204.	-6178.	5454.	-4468.	565.5	100529.
	38.3	[illegible]	51.1	12445.	-2453.	-1204.	-6178.	5454.	-4468.	740.	98332.
	0.162		44.3	14024.	-3200.	-1303.	-6643.	5876.	-4842.	0.253	1764.2
	97.	0.2193	50.1		-77.41						1933.5
	685.4	0.9228	44.7	1.7657	-0.4255	-0.1735	-0.1693	0.0244	-0.0200		0.01249

TABLE V.— Continued.

RUN 23 TEST 525

PT	VKTS / QPSF / MTUN / TEMP / VSNDKTS	ALPHA,D / PSI / / RHO*100 / RHOF	INA / PED POS / LONG STK / LAT STK / P34 LEV	LIFT,G / LIFT / LIFT/G / / CL	DRAG,G / DRAG / DRAG/G / FE / CD	SIDE,G / SIDE / SIDE/G / / CY	PITCH,G / PITCH / PITCH/G / / CM	YAW,G / YAW / YAW/G / / CN	ROLL,G / ROLL / ROLL/G / / CROLL	RPM / OMEGA R / V/OR	THRUST,G / THRUST / HP / HP/Q / CP/SIGMA
11	110.7	8.0	0.0	13050.	-2988.	-680.	-5652.	4087.	-3116.	565.2	49790.
	38.3	-2.0	51.3	13050.	-2988.	-680.	-5652.	4087.	-3116.	740.	49409.
	0.162		44.4	14142.	-3230.	-737.	-7208.	4424.	-3376.	0.252	1717.9
	97.	0.2193	50.0		-78.08						1920.7
	685.9	0.9220	44.6	1.0890	-0.2313	-0.0982	-0.1028	0.0184	-0.0149		0.012566
12	111.0	8.0	0.0	13015.	-2949.	485.	-4904.	-2425.	1082.	565.5	49277.
	38.4	2.0	51.3	13015.	-2949.	485.	-4904.	-2425.	1082.	740.	49507.
	0.162		44.5	14135.	-3205.	527.	-7569.	-2633.	1427.	0.253	1774.4
	95.	0.2184	50.1		-76.81						1927.1
	686.3	0.9208	44.6	1.0724	-0.2244	0.0048	-0.1908	-0.0120	0.0075		0.012567
13	110.9	8.0	0.0	13040.	-2869.	968.	-6040.	-4824.	2448.	565.0	47734.
	38.3	4.0	51.2	13040.	-2869.	968.	-6040.	-4824.	2448.	740.	48607.
	0.162		49.4	14152.	-3115.	1051.	-7218.	-5422.	3244.	0.254	1752.1
	98.	0.2184	50.1		-74.80						1915.5
	686.0	0.9208	44.6	1.0799	-0.2136	0.0139	-0.1823	-0.0267	0.0132		0.012467
14	111.1	8.0	0.0	12930.	-2829.	2011.	-5300.	-7640.	6172.	565.0	47840.
	38.4	6.0	51.1	12930.	-2829.	2011.	-5300.	-7640.	6172.	740.	49672.
	0.162		49.5	14068.	-3079.	2194.	-5766.	-8312.	6715.	0.254	1771.2
	99.	0.2185	50.2		-73.54						1927.0
	686.7	0.9191	44.6	1.0595	-0.2058	0.2842	-0.1456	-0.0342	0.0276		0.012545
15	110.3	8.0	0.0	12545.	-2966.	-114.	-2384.	511.	-165.	565.3	49112.
	37.8	0.0	51.2	12545.	-2966.	-114.	-2384.	511.	-165.	740.	47045.
	0.161		34.5	13675.	-3233.	-124.	-2599.	557.	-180.	0.252	1752.4
	100.	0.2181	44.9		-75.50						1910.2
	687.3	0.9174	44.6	1.0343	-0.2337	-0.0767	-0.0663	0.0023	-0.0007		0.012466
16	110.2	8.0	0.0	12966.	-2910.	-60.	-12504.	-37.	-1741.	565.0	49232.
	37.7	0.0	51.1	12980.	-2910.	-63.	-12504.	-37.	-1745.	741.	47276.
	0.161		59.8	14149.	-3172.	-65.	-13630.	-40.	-1903.	0.251	1755.6
	100.	0.2191	49.8		-77.11						1915.7
	687.3	0.9174	44.6	1.0903	-0.2260	-0.0058	-0.3454	-0.0002	-0.0079		0.012497
17	110.5	8.0	0.0	12730.	-2967.	-221.	-6408.	3585.	5610.	565.4	49179.
	37.9	0.0	50.4	12730.	-2967.	-221.	-6408.	3585.	5610.	740.	[illegible]
	0.161		50.0	13877.	-3234.	-241.	-6985.	3906.	6124.	0.252	1756.7
	100.	0.2180	69.9		-76.23						1915.0
	687.5	0.9173	44.6	1.0545	-0.2322	-0.0322	-0.1776	0.0162	0.0254		0.012484
18	110.6	8.0	0.0	12270.	-2959.	-254.	-4901.	6393.	12745.	565.6	50345.
	37.9	0.0	50.4	12270.	-2959.	-254.	-4901.	6393.	12745.	740.	[illegible]
	0.161		50.2	13400.	-3231.	-277.	-5418.	6981.	13918.	0.252	1745.9
	101.	0.2177	64.5		-76.52						1906.7
	687.9	0.9157	44.6	[illegible]	-0.2511	-0.0370	-0.1375	0.0289	0.0577		0.012426
19	110.6	8.0	0.0	12800.	-2858.	-356.	-7770.	6312.	-705.	565.3	49054.
	37.9	0.0	69.9	12800.	-2858.	-358.	-7770.	6312.	-705.	741.	45794.
	0.161		50.0	13979.	-3121.	-391.	-8485.	6894.	-770.	0.252	1728.1
	101.	0.2176	49.8		-75.31						1885.0
	687.9	0.9157	44.6	1.0635	-0.2161	-0.0521	-0.2153	0.0288	-0.0032		0.012304
20	110.5	8.0	0.0	12735.	-2859.	-517.	-6150.	11089.	-2440.	565.4	49454.
	37.8	0.0	49.5	12735.	-2859.	-517.	-6150.	11089.	-2440.	741.	47605.
	0.161		50.1	13932.	-3125.	-566.	-6729.	12131.	-2670.	0.252	1751.3
	102.	0.2173	49.9		-75.00						1900.6
	688.5	0.9141	44.6	1.0627	-0.2386	-0.0756	-0.1711	0.0504	-0.0111		0.012576
21	110.4	8.0	0.0	12750.	-2820.	-495.	-6034.	10537.	-1872.	565.4	45475.
	37.7	0.0	89.6	12750.	-2820.	-478.	-6034.	10537.	-1872.	740.	45521.
	0.161		50.0	13948.	-3085.	-545.	-7257.	11527.	-2048.	0.252	1717.5
	102.	0.2173	49.7		-74.72						1879.3
	688.5	0.9141	44.6	1.0666	-0.2128	-0.0729	-0.1684	0.0450	-0.0085		0.012259
22	150.0	-1.0	0.0	12030.	753.	-94.	15782.	3049.	-684.	566.5	56021.
	75.7	0.0	50.2	12030.	753.	-94.	15782.	3049.	-684.	742.	53475.
	0.230		50.0	13410.	841.	-105.	17622.	3405.	-764.	0.360	1115.4
	106.	0.2129	49.4		9.45						1245.4
	690.9	0.8950	30.2	0.6765	0.0553	-0.0064	0.2192	0.0064	-0.0010		0.008017
23	150.3	-5.0	0.0	6680.	315.	-114.	14915.	3600.	1404.	565.3	56287.
	75.7	0.0	50.2	6683.	315.	-114.	14916.	3600.	1404.	740.	53235.
	0.230		50.0	7465.	413.	-125.	16714.	4101.	1640.	0.361	1072.0
	106.	0.2121	49.7		10.77						1201.2
	692.2	0.8924	30.2	0.4975	0.0595	-0.0083	0.2071	0.0063	0.0033		0.007841
24	150.7	-3.0	0.0	9420.	704.	-136.	15545.	2614.	2007.	566.	56244.
	75.8	0.0	50.4	9420.	704.	-136.	15545.	2614.	2007.	741.	53316.
	0.230		50.1	10575.	801.	-153.	17451.	2433.	2253.	0.361	1091.9
	109.	0.2117	44.6		10.13						1221.9
	692.8	0.8908	30.1	0.6855	0.0559	-0.0094	0.2152	0.0059	0.0045		0.007985
25	150.7	-1.0	0.0	12020.	802.	-138.	14834.	2903.	955.	565.7	57044.
	75.9	0.0	50.3	12020.	802.	-138.	14834.	2903.	955.	741.	53442.
	0.230		50.1	13517.	902.	-155.	16681.	3265.	1112.	0.362	[illegible]
	110.	0.2114	44.7		10.57						[illegible]
	693.4	0.8893	30.2	0.8755	0.0594	[illegible]	0.2156	0.0066	0.0027		0.007940
26	150.8	1.0	0.0	[illegible]	414.	-122.	13355.	3052.	1124.	566.	[illegible]
	75.6	0.0	50.5	[illegible]	414.	-122.	13355.	3052.	1124.	741.	[illegible]
	0.230		50.1	[illegible]	1035.	-137.	14706.	3434.	1265.	0.362	[illegible]
	111.	0.2110	[illegible]		12.12						[illegible]
	694.0	0.8877	30.1	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]		[illegible]
27	150.6	3.0	0.0	17390.	1112.	-134.	9731.	2814.	2307.	565.8	58425.
	75.7	0.0	50.5	17390.	1112.	-134.	9731.	2810.	2307.	741.	[illegible]
	0.230		50.1	19621.	1255.	-151.	[illegible]	3171.	2671.	0.362	[illegible]
	112.	0.2106	49.8		[illegible]						[illegible]
	694.6	0.8862	30.1	[illegible]	[illegible]	[illegible]	0.1352	[illegible]	[illegible]		[illegible]

TABLE V.— Continued.

RUN 26 TEST 525

PT	VELS QPSF MTUN TEMP VSNDFTS	ALPHA,C PSI RHO*1000 RHOR	[illegible] PER POS LONG STK LAT STK POW LEV	LIFT,LB LIFT LIFT/6 CL	DRAG,LB DRAG DRAG/6 FE CD	SIDE,LB SIDE SIDE/6 CY	PITCH,LB PITCH PITCH/6 CM	YAW,LB YAW YAW/6 CN	ROLL,LB ROLL ROLL/6 CROLL	RPM OMEGAR V/OR	SHAFT,LB SHAFT,LB HP HP/6 CP/SIGMA
3	60.0	-5.0	40.0	12595.	216.	-77.	-996.	-92.	-1401.	500.1	44462.
	11.5	0.0	49.9	12595.	216.	-77.	-996.	-92.	-1401.	741.	43271.
	0.089		59.2	13082.	224.	-81.	-1051.	-97.	-1478.	0.137	787.5
	86.	0.2252	49.3		18.71						851.1
	678.6	0.9475	16.1	5.9317	0.1037	-0.0368	-0.0997	-0.0014	-0.0208		0.005402
4	80.2	-8.0	40.0	12560.	253.	-115.	-1093.	-677.	-1647.	500.2	54764.
	20.5	0.0	49.9	12560.	253.	-115.	-1093.	-677.	-1647.	741.	54162.
	0.118		66.6	13133.	269.	-125.	-1162.	-720.	-1750.	0.183	976.5
	88.	0.2237	49.2		12.34						1039.7
	679.9	0.9412	21.2	3.3295	0.0682	-0.0318	-0.0561	-0.0057	-0.0138		0.006755
5	80.2	0.0	40.0	15950.	1685.	-114.	-5767.	-154.	-1162.	500.0	41905.
	20.5	0.0	49.9	15950.	1685.	-114.	-5767.	-154.	-1162.	741.	41827.
	0.118		69.2	16958.	1794.	-121.	-6134.	-164.	-1173.	0.183	752.5
	89.	0.2233	49.4		82.35						801.0
	680.5	0.9395	17.1	4.5344	0.4550	-0.0306	-0.2465	-0.0013	-0.0092		0.005226
6	80.5	0.0	75.0	10525.	134.	-67.	-2126.	-451.	-103.	500.0	42744.
	20.6	0.0	49.9	10525.	134.	-67.	-2126.	-451.	-103.	741.	42256.
	0.119		64.5	10991.	143.	-71.	-2265.	-480.	-104.	0.183	763.3
	89.	0.2233	49.2		6.50						812.5
	680.5	0.9394	17.1	2.7651	0.0359	-0.0179	-0.1364	-0.0036	-0.0009		0.005265
7	79.8	0.0	60.0	7965.	-372.	-46.	-457.	-160.	337.	500.2	41506.
	20.2	0.0	49.9	7965.	-372.	-46.	-457.	-160.	337.	741.	42510.
	0.117		64.1	8491.	-391.	-49.	-487.	-170.	359.	0.182	754.7
	90.	0.2224	44.2		-16.40						844.6
	681.1	0.9376	17.2	2.1762	-0.1016	-0.0126	-0.0238	-0.0014	0.0029		0.005228
8	80.0	0.0	45.0	7115.	-712.	-85.	-932.	-15.	-225.	500.2	40407.
	20.3	0.0	69.8	7115.	-712.	-85.	-932.	-15.	-225.	741.	40734.
	0.118		64.3	7567.	-754.	-91.	-994.	-16.	-239.	0.182	733.5
	90.	0.2224	44.4		-35.30						782.1
	681.1	0.9378	17.1	1.9304	-0.1936	-0.0231	-0.0482	-0.0001	-0.0019		0.005080
9	80.2	0.0	30.0	6055.	-1184.	-55.	2776.	-348.	-258.	595.7	41266.
	20.4	0.0	44.9	6055.	-1184.	-55.	2776.	-348.	-256.	740.	41281.
	0.118		55.0	6451.	-1266.	-59.	2965.	-371.	-273.	0.183	741.1
	90.	0.2224	49.1		-58.25						790.3
	681.1	0.9378	17.3	1.6374	-0.3216	-0.0149	0.1431	-0.0029	-0.0022		0.005147
10	79.9	-4.0	60.0	7250.	-1257.	-67.	1220.	-1675.	-2794.	565.6	50794.
	20.2	0.0	49.9	7250.	-1257.	-67.	1220.	-1675.	-2794.	740.	46717.
	0.117		64.7	7725.	-1124.	-72.	1304.	-1796.	-2984.	0.182	893.6
	91.	0.2225	49.3		-52.21						955.9
	681.7	0.9361	18.5	1.9730	-0.2864	-0.0183	0.0634	-0.0142	-0.0237		0.006217
11	80.2	-2.0	60.0	8005.	-895.	-80.	-544.	-1060.	-1801.	500.0	53195.
	20.3	0.0	50.0	8005.	-895.	-80.	-544.	-1060.	-1801.	741.	47475.
	0.118		64.7	8567.	-956.	-85.	-580.	-1070.	-1927.	0.183	861.7
	92.	0.2221	49.2		-44.00						943.6
	682.3	0.9344	16.6	2.1742	-0.2431	-0.0217	-0.0264	-0.0084	-0.0152		0.006134
12	80.1	2.0	60.0	9460.	-674.	-56.	-2968.	-1361.	-2352.	500.0	49615.
	20.3	0.0	49.8	9460.	-674.	-56.	-2968.	-1361.	-2352.	741.	48053.
	0.118		64.7	10145.	-721.	-54.	-3176.	-1392.	-2517.	0.182	877.1
	92.	0.2221	49.3		-33.21						932.7
	682.3	0.9344	18.5	2.5809	-0.1835	-0.0156	-0.1536	-0.0110	-0.0199		0.006104
13	80.1	2.0	60.0	9575.	-662.	-56.	-176.	-1215.	-2537.	500.1	49835.
	20.1	0.0	49.7	9575.	-662.	-56.	-176.	-1215.	-2537.	741.	48229.
	0.118		59.6	10247.	-706.	-60.	-188.	-1290.	-2740.	0.182	880.6
	92.	0.2221	49.2		-32.02						942.7
	682.3	0.9344	16.5	2.6066	-0.1802	-0.0152	-0.0091	-0.0102	-0.0221		0.006126
14	80.0	5.0	60.0	9585.	-589.	-65.	-978.	-1171.	-2441.	500.2	49496.
	20.2	0.0	50.0	9685.	-589.	-65.	-978.	-1171.	-2441.	741.	47643.
	0.117		59.7	10596.	-631.	-70.	-1044.	-1250.	-3202.	0.182	872.6
	93.	0.2217	49.2		-29.13						935.5
	683.0	0.9326	18.5	2.7008	-0.1609	-0.0178	-0.0509	-0.0099	-0.0254		0.006076
15	100.8	-2.0	60.0	9725.	84.	-141.	-894.	640.	-560.	500.5	54369.
	31.9	0.0	49.8	9725.	84.	-141.	-894.	640.	-560.	741.	54904.
	0.148		64.9	10407.	91.	-152.	-965.	675.	-711.	0.224	982.5
	94.	0.2204	49.3		2.63						1059.3
	683.6	0.9273	23.3	1.6847	0.0145	-0.0244	-0.0296	0.0024	-0.0036		0.006871
16	100.7	0.0	60.0	10755.	205.	-127.	-2664.	-516.	-1266.	500.1	55606.
	31.6	0.0	50.0	10755.	205.	-127.	-2664.	-516.	-1266.	741.	54221.
	0.148		64.4	11547.	221.	-137.	-2595.	-556.	-1366.	0.224	986.6
	94.	0.2204	49.3		6.44						1063.6
	683.6	0.9274	23.3	1.8659	0.0355	-0.0221	-0.0887	-0.0024	-0.0068		0.006912
17	120.1	-4.0	60.0	9905.	168.	-176.	3264.	31.	-731.	500.5	50808.
	44.9	0.0	49.9	9905.	168.	-176.	3264.	31.	-731.	742.	54607.
	0.176		73.7	10769.	170.	-194.	3570.	32.	-795.	0.273	1003.3
	96.	0.2186	49.4		25.99						1090.6
	684.8	0.9198	23.3	1.2175	0.0436	-0.0219	0.0766	0.0000	-0.0028		0.007072
18	120.4	-2.0	60.0	11535.	1346.	-209.	-196.	-704.	-1956.	505.6	50627.
	45.0	0.0	50.0	11535.	1346.	-209.	-196.	-708.	-1956.	742.	53499.
	0.176		73.7	12564.	1455.	-220.	-215.	-762.	-2103.	0.274	980.2
	97.	0.2182	49.2		24.43						1076.3
	685.4	0.9161	23.3	1.4150	0.1654	-0.0256	-0.0048	-0.0027	-0.0076		0.006978
19	120.1	-1.0	60.0	12220.	1449.	-216.	-2274.	45.	-2171.	500.4	50327.
	44.6	0.0	49.4	12126.	1449.	-216.	-2270.	45.	-2171.	741.	52972.
	0.176		73.7	13223.	1561.	-229.	-2481.	49.	-2359.	0.273	962.2
	98.	0.2179	49.4		32.35						1071.7
	686.0	0.9166	23.0	1.4952	0.1786	-0.0269	-0.0534	0.0002	-0.0081		0.006954
20	120.1	0.0	60.0	12995.	1545.	-196.	-3421.	-499.	-2959.	500.3	53508.
	44.4	0.0	49.6	12995.	1545.	-196.	-3421.	-499.	-2959.	742.	52992.
	0.176		73.8	14124.	1669.	-214.	-3732.	-544.	-3261.	0.273	965.9
	98.	0.2179	49.5		34.56						1073.4
	686.0	0.9166	23.6	1.5969	0.1910	-0.0242	-0.0843	-0.0019	-0.0165		0.006948

ORIGINAL PAGE IS
OF POOR QUALITY

TABLE V.— Continued.

RUN 24 TEST 525

PT	VAIS / QPSF / MACH / TEMP / VSNDKTS	ALPHA / PSI / — / RHOALT / RHOR	ISA / PER PCS / LONG STK / LAT STK / PDAL EF	LIFT / LIFT / LIFT/Q / — / CL	DRAG / DRAG / DRAG/Q / FE / CD	SIDE / SIDE / SIDE/Q / — / CY	PITCH / PITCH / PITCH/Q / — / CM	YAW / YAW / YAW/Q / — / CN	ROLL / ROLL / ROLL/Q / — / CROLL	RPM / MU / VTIP	SHAFT / SHAFT / HP / — / CP/SIGMA
21	120.6	0.0	56.[illegible]	8800.	439.	-167.	-6747.	139.	-1400.	500.4	50765.
	45.1	0.0	44.5	6660.	439.	-167.	-6747.	138.	-1400.	741.	525[illegible]
	0.176		73.9	5471.	1025.	-182.	-5180.	131.	-1327.	0.275	982.[illegible]
	96.	0.2178	44.6		20.51						1072.4
	688.3	0.9164	23.3	1.0020	0.1149	-0.0204	-0.1106	0.0005	-0.0055		0.00694
22	99.6	-5.0	75.0	10175.	284.	-194.	5094.	-597.	-2865.	590.[illegible]	58[illegible]
	30.9	0.0	50.0	10175.	284.	-194.	5094.	-597.	-2865.	741.	56363.
	0.145		84.6	11066.	309.	-211.	5538.	-652.	-3085.	0.227	1032.7
	94.	0.2183	49.5		9.19						1125.3
	686.7	0.9195	25.1	1.8203	0.0508	-0.0347	0.1731	-0.0033	-0.0156		0.00753
23	99.7	-3.0	75.0	11735.	43.	-167.	2441.	-957.	-3612.	550.3	59595.
	30.9	0.0	49.5	11735.	43.	-167.	2441.	-957.	-3612.	741.	59444.
	0.145		64.5	12787.	47.	-182.	2660.	-1043.	-3935.	0.227	1046.8
	100.	0.2181	49.3		13.46						1134.3
	687.3	0.9177	25.1	2.1007	0.0772	-0.0299	0.0832	-0.0053	-0.0201		0.00753
24	99.6	-2.0	75.0	12725.	477.	-173.	2071.	-569.	-5444.	550.5	59942.
	30.8	0.0	50.0	12725.	477.	-173.	2071.	-569.	-5444.	742.	57[illegible]
	0.145		54.5	13866.	520.	-159.	2257.	-620.	-5758.	0.227	1038.3
	100.	0.2181	49.4		15.46						1153.2
	687.3	0.9177	25.1	2.2614	0.0855	-0.0310	0.0707	-0.0032	-0.0142		0.00747
25	99.7	-2.0	75.0	12730.	535.	-1254.	2844.	3851.	-7947.	560.2	59922.
	30.9	-4.0	50.0	12730.	535.	-1254.	2844.	3851.	-7947.	741.	57[illegible]
	0.145		64.4	13872.	583.	-1144.	3044.	4196.	-8665.	0.227	1051.[illegible]
	100.	0.2181	49.5		17.32						1140.[illegible]
	687.3	0.9177	25.1	2.2770	0.0957	-0.1365	0.0960	0.0214	-0.0443		0.00744
26	99.8	-2.0	75.0	12690.	514.	-600.	2369.	1059.	-5414.	560.2	59502.
	31.0	-2.0	49.6	12690.	514.	-600.	2369.	1059.	-5414.	741.	57520.
	0.146		64.3	13826.	560.	-654.	2534.	1154.	-5900.	0.227	1054.5
	100.	0.2181	49.3		16.60						1149.2
	687.3	0.9177	25.1	2.2567	0.0917	-0.1072	0.0811	0.0059	-0.0300		0.00740
27	99.7	-2.0	75.0	12670.	506.	295.	2144.	-3153.	-700.	560.4	59702.
	30.9	2.0	50.0	12670.	506.	295.	2144.	-3153.	-700.	741.	56164.
	0.146		64.2	13806.	551.	321.	2337.	-3436.	-763.	0.227	1054.2
	100.	0.2181	49.5		16.37						1154.2
	687.3	0.9177	25.1	2.2648	0.0904	0.0527	0.0729	-0.0175	-0.0039		0.00744
28	99.8	-2.0	75.0	12785.	511.	779.	2838.	-5269.	2218.	560.4	59165.
	30.9	4.0	50.0	12785.	511.	779.	2838.	-5269.	2218.	741.	57890.
	0.146		64.3	13932.	557.	849.	3093.	-5763.	2417.	0.227	1052.1
	100.	0.2181	49.4		16.52						1140.5
	687.3	0.9177	25.1	2.2830	0.0913	0.1391	0.0965	-0.0294	0.0123		0.00743
29	99.6	-2.0	75.0	12745.	604.	1645.	2806.	-9238.	5059.	560.5	58744.
	30.8	8.0	49.8	12745.	604.	1645.	2806.	-9238.	5059.	742.	57734.
	0.145		64.4	13887.	658.	1792.	3125.	-10066.	5513.	0.227	1046.9
	100.	0.2181	49.5		18.62						1140.7
	687.3	0.9177	25.1	2.2867	0.1364	0.2951	0.0979	-0.0515	0.0282		0.00739
30	99.7	-2.0	75.0	12610.	731.	2445.	4798.	-12064.	9791.	560.3	59165.
	30.9	12.0	49.9	12610.	731.	2445.	4798.	-12064.	9791.	741.	585[illegible]
	0.145		64.2	13741.	797.	2664.	5228.	-13151.	10669.	0.227	1057.4
	100.	0.2181	49.4		23.07						1152.2
	687.3	0.9177	25.1	2.2556	0.1306	0.4373	0.1633	-0.0671	0.0544		0.00743
31	99.8	-2.0	75.0	13380.	667.	-178.	9683.	-517.	-3610.	560.5	59002.
	30.9	0.0	49.9	13380.	667.	-178.	9683.	-517.	-3610.	742.	57734.
	0.145		54.8	14606.	726.	-194.	10572.	-564.	-3940.	0.227	1055.4
	101.	0.2177	49.5		21.59						1152.1
	687.9	0.9161	25.1	2.3913	0.1193	-0.0316	0.3296	-0.0029	-0.0201		0.00747
32	99.8	-2.0	75.0	12020.	378.	-159.	-5281.	-471.	-2493.	560.8	59493.
	30.9	0.0	50.0	12020.	378.	-159.	-5281.	-471.	-2493.	742.	57715.
	0.145		73.7	13121.	4[illegible]3.	-174.	-5765.	-451.	-27[illegible]1.	0.227	1053.7
	101.	0.2177	49.6		12.49						1152.2
	687.9	0.9161	25.1	2.1500	0.0676	-0.0284	-0.1798	-0.0044	-0.0138		0.00745
33	99.7	-2.0	75.0	12505.	528.	-1[illegible].	2025.	49.	20110.	560.8	50794.
	30.8	0.0	49.8	12505.	528.	-1[illegible].	2025.	49.	20110.	742.	60517.
	0.145		63.6	13650.	574.	-129.	2211.	54.	21980.	0.227	1053.2
	101.	0.2177	64.4		17.07						1149.[illegible]
	687.9	0.9161	25.1	2.2419	0.0943	-0.0212	0.0691	0.0003	0.1121		0.00744
34	99.7	-2.0	75.0	12020.	608.	-83.	3367.	1067.	41512.	560.3	41800.
	30.8	0.0	49.9	12020.	606.	-65.	3367.	1067.	41542.	741.	75194.
	0.145		63.5	13321.	662.	-91.	3675.	1167.	45347.	0.227	1051.4
	101.	0.2177	84.4		19.65						1147.7
	687.9	0.9161	25.1	2.1533	0.1066	-0.0149	0.1145	0.0[illegible]	0.2115		0.00745
35	99.9	-2.0	75.0	12660.	493.	-420.	2232.	6225.	-5400.	555.1	60242.
	31.0	0.0	70.0	12660.	493.	-420.	2232.	6225.	-5400.	741.	58769.
	0.146		63.9	13842.	538.	-459.	2437.	6793.	-5895.	0.226	1068.9
	101.	0.2177	50.6		15.92						1160.8
	687.9	0.9163	25.1	2.2629	0.0880	-0.0750	0.0758	0.0345	-0.0300		0.00758
36	99.6	-2.0	75.0	12740.	521.	-593.	2574.	10940.	-10271.	555.5	60102.
	31.0	0.0	84.6	12740.	521.	-593.	2574.	10940.	-10271.	742.	58457.
	0.145		63.9	13906.	569.	-647.	3138.	11943.	-11213.	0.227	1068.[illegible]
	101.	0.2177	50.0		16.[illegible]5						1163.[illegible]
	687.9	0.9160	25.1	2.2736	0.0930	-0.1058	0.0970	[illegible]	[illegible]		0.00759
37	139.4	-5.0	30.0	9485.	390.	-220.	8016.	-927.	-2434.	560.4	81131.
	59.5	0.0	44.9	9485.	390.	-220.	8016.	-927.	-2434.	742.	81316.
	0.203		76.3	10495.	432.	-243.	8857.	-1026.	-3209.	0.317	1182.1
	103.	0.2148	56.0		6.50						1309.2
	689.1	0.9037	26.7	1.8811	0.0362	-0.0204	0.1593	-0.0027	-0.0045		[illegible]
38	139.3	-3.0	30.0	13515.	524.	-171.	3310.	-1410.	-1741.	560.5	[illegible]
	59.3	0.0	44.4	13515.	524.	-171.	3310.	-1410.	-1741.	742.	[illegible]
	0.203		74.5	14976.	580.	-190.	3675.	-1565.	-1930.	0.317	1199.3
	104.	0.2144	49.4		8.45						1314.[illegible]
	689.7	0.9022	26.5	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]		[illegible]

TABLE V.— Continued.

RUN 24 TEST 525

PT	VELO QPSF MTUN TEMP VSOUNDS	ALPS,C PSI RHO/RHOO RHOR	INA REC POS LONG STR LAT STR POW LEV	LIFT,U LIFT LIFT/Q CL	DRAG,U DRAG DRAG/Q FE CD	SIDE,U SIDE SIDE/Q CY	PITCH,U PITCH PITCH/Q CM	YAW,U YAW YAW/Q CN	ROLL,U ROLL ROLL/Q CROLL	RPM OMEGA R V/OR	HMAST Q CMAST Q HP HP/Q CP/SIGMA
39	139.3 59.3 0.203 104. 639.7	-1.0 0.0 0.2144 0.9022	30.0 50.0 74.5 44.4 28.6	13850. 13850. 15352. 1.2905	754. 754. 816. 12.72 0.0723	-186. -186. -206. -0.0173	-437. -437. -485. -0.0074	-1153. -1153. -1276. -0.0033	-2645. -2645. -2935. -0.0077	566.5 742. 0.317	64430. 62067. 1362.5 1310.7 0.008484
40	139.5 59.3 0.203 105. 640.3	0.0 0.0 0.2141 0.9006	30.0 50.0 74.5 50.0 26.5	15625. 15625. 17572. 1.4739	1082. 1082. 1201. 15.24 0.1058	-282. -282. -313. -0.0263	-2684. -2684. -3245. -0.2052	-1053. -1053. -1164. -0.0050	-4137. -4137. -4594. -0.0126	566.5 742. 0.317	64436. 62653. 1327.2 1316.3 0.008556
41	139.5 59.2 0.203 106. [illegible]	3.0 0.0 0.2137 0.8990	30.0 50.0 74.5 50.0 26.5	17980. 17980. 20000. 1.6772	1395. 1395. 1552. 23.55 0.1301	+269. +269. +299. -0.0251	-6864. -6864. -7635. -0.1218	-1353. -1353. -1505. -0.0039	-4686. -4686. -5035. -0.0042	566.1 741. 0.318	64689. 61599. 1174.1 1311.5 0.008524

RUN 26 TEST 525

PT	VELO QPSF MTUN TEMP VSOUNDS	ALPS,C PSI RHO/RHOO RHOR	INA REC POS LONG STR LAT STR POW LEV	LIFT,U LIFT LIFT/Q CL	DRAG,U DRAG DRAG/Q FE CD	SIDE,U SIDE SIDE/Q CY	PITCH,U PITCH PITCH/Q CM	YAW,U YAW YAW/Q CN	ROLL,U ROLL ROLL/Q CROLL	RPM OMEGA R V/OR	HMAST Q CMAST Q HP HP/Q CP/SIGMA
3	39.6 5.2 0.054 74. 671.1	-2.0 0.0 0.2312 0.9725	40.0 49.5 54.9 50.1 20.5	12995. 12995. 13362. 13.7890	65. 65. 70. 13.06 0.3722	-5. -5. -5. -0.0053	-639. -639. -657. -0.1290	123. 123. 127. 0.0041	-1776. -1776. -1826. -0.0567	565.3 740. 0.041	56027. 55605. 1001.8 1030.[illegible] 0.006723
4	60.1 11.6 0.049 76. 672.0	-5.0 0.0 0.2298 0.9667	40.0 49.5 61.0 50.1 18.5	13050. 13050. 13444. 6.1039	124. 124. 133. 10.92 0.0603	-96. -96. -101. -0.0458	-3113. -3113. -3234. -0.2787	52. 52. 54. 0.0018	-2042. -2042. -2112. -0.0297	565.1 740. 0.137	48757. 48678. 873.0 905.6 0.005905
5	79.6 20.6 0.118 76. 673.6	-8.0 0.0 0.2282 0.9603	40.0 49.5 64.0 50.1 21.4	12730. 12730. 13257. 3.4132	225. 225. 234. 10.42 0.0603	-96. -96. -102. -0.0263	-2653. -2653. -2763. +0.1354	-949. -949. -982. +0.0079	-1451. -1451. -1511. -0.0121	565.3 740. 0.162	57525. 57520. 1031.8 1074.5 0.007014
6	139.6 61.3 0.207 87. 679.2	-1.0 -4.0 0.2213 0.9311	30.0 49.5 74.6 50.6 23.3	14345. 14345. 15906. 1.2900	1295. 1295. 1391. 21.06 0.1165	-2004. -2004. -2152. -0.1802	1133. 1133. 1216. 0.0194	7330. 7330. 7854. 0.0199	-11665. -11665. -12544. -0.0327	565.6 740. 0.318	64150. 50942. 1077.5 1157.3 0.007541
7	139.6 61.3 0.206 95. 679.9	-1.0 -2.0 0.2249 0.9295	30.0 49.6 74.5 50.0 23.2	14000. 14000. 15443. 1.2773	1237. 1237. 1331. 20.17 0.1114	-1154. -1154. -1242. -0.1040	1. 1. 1. 0.0000	4116. 4118. 4431. 0.0115	-7002. -7002. -7533. -0.0196	565.3 740. 0.318	63477. 55529. 1071.4 1153.2 0.007527
8	140.0 61.5 0.207 85. 680.5	-1.0 2.0 0.2205 0.9277	30.0 49.5 74.4 50.8 23.3	14420. 14420. 15544. 1.2944	1224. 1220. 1319. 19.64 0.1095	645. 645. 695. 0.0579	1155. 1159. 1249. 0.0196	-2777. -2777. -2995. -0.0077	5238. 5238. 5647. 0.0146	564.9 739. 0.320	55268. 63824. 1067.4 1150.6 0.007526
9	140.6 61.4 0.207 90. 681.[illegible]	-1.0 0.0 0.2201 0.9260	30.0 49.4 74.4 50.8 23.3	14375. 14375. 15523. 1.2927	1295. 1295. 1402. 21.13 0.1167	3387. 3387. 3659. 0.3046	2068. 2068. 2255. 0.0357	3282. 3282. 3544. 0.0092	-47620. -47620. -50775. -0.1314	565.3 740. 0.319	56594. 60383. 1044.3 1133.1 0.007344
10	139.9 61.3 0.206 91. 681.7	-1.0 5.0 0.2147 0.9244	30.0 49.4 74.6 50.4 23.3	14145. 14145. 15342. 1.2752	1464. 1464. 1564. 25.59 0.1326	6524. 6524. 7058. 0.5882	[illegible] [illegible] [illegible] [illegible]	[illegible] [illegible] [illegible] [illegible]	[illegible] [illegible] [illegible] [illegible]	565.5 740. 0.319	45284. 73524. 1048.2 1133.9 0.007391
11	140.2 61.4 0.206 92. 682.3	-1.0 0.0 0.2193 0.9227	30.0 49.5 64.6 50.7 23.3	13875. 13875. 15034. 1.2486	1260. 1260. 1366. 20.52 0.1134	-220. -226. -245. -0.0205	16375. 16375. 17245. 0.1377	362. 362. 414. 0.0011	[illegible] [illegible] -36. 0.0011	565.8 741. 0.319	56664. 56944. 1056.0 1144.5 0.007444
12	140.2 61.4 0.206 92. 682.3	-1.0 0.0 0.2193 0.9227	30.0 49.5 64.1 51.5 23.3	14580. 14580. 15601. 1.3126	1263. 1263. 1390. 20.91 0.1155	-223. -223. -242. -0.0201	-4702. -4702. [illegible] [illegible]	[illegible] [illegible] [illegible] 0.0025	[illegible] [illegible] 653. [illegible]	565.4 740. 0.320	58324. 58730. 1053.1 1135.0 0.007424
13	140.4 61.4 0.207 93. 683.0	-1.0 0.0 0.2189 0.9210	30.0 49.2 73.5 70.1 23.4	14160. 14160. 15374. 1.2734	[illegible] [illegible] 1462. 22.01 0.1141	-216. -216. [illegible] [illegible]	[illegible] [illegible] [illegible] [illegible]	10546. 10546. [illegible] [illegible]	17088. 17088. 18555. [illegible]	565.5 [illegible] 0.320	54194. 75242. 1026.7 1114.7 0.007269
14	140.5 61.5 0.207 93. 683.0	-1.0 0.0 0.2189 0.9210	30.0 49.6 73.7 44.4 23.3	13570. 13570. 14734. 1.2141	1350. 1350. 1472. 22.03 0.1207	-399. -399. -422. [illegible]	1397. 1397. 1517. 0.0234	22184. 22184. [illegible] [illegible]	31249. 31249. [illegible] 0.1041	565.5 740. 0.320	74215. 95092. 1023.4 1111.5 0.007264

ORIGINAL PAGE IS OF POOR QUALITY

TABLE V.— Continued.

RUN 26 TEST 525

PT	V KTS / Q PSF / MACH / TEMP / V KNOTS	ALPHA / PSI / RHO / RHOR	PED POS / LONG STK / LAT STK / PWR LEV	LIFT / LIFT/Q / CL	DRAG / DRAG/Q / FE / CD	SIDE / SIDE/Q / CY	PITCH / PITCH/Q / CM	YAW / YAW/Q / CN	ROLL / ROLL/Q / CROLL	RPM / ADVANCE RATIO	HP / HP/Q / CP/SIGMA
15	140.2	-1.0	50.0	14245.	1328.	+554.	1200.	7643.	-4503.	505.4	51520.
	61.3	0.0	65.6	14245.	1328.	+554.	1200.	7643.	-4503.	742.	54143.
	0.206		73.9	15486.	1440.	+601.	1340.	8298.	-4889.	0.320	1035.9
	93.	0.2189	50.0		21.67						1124.7
	683.0	0.9210	23.3	1.2833	0.1195	+0.0409	0.0219	0.0214	-0.0120		0.007535
16	140.3	-1.0	50.0	14305.	1339.	+644.	750.	14512.	-7276.	505.6	62710.
	61.3	0.0	84.0	14305.	1339.	+644.	750.	14512.	-7276.	740.	52525.
	0.206		73.9	15559.	1456.	+425.	800.	15784.	-7910.	0.320	1054.1
	94.	0.2165	50.0		21.64						1124.6
	683.6	0.9195	23.5	1.2892	0.1207	+0.0705	0.0126	0.0403	-0.0204		0.007311
17	100.7	0.0	45.0	15795.	-2425.	-80.	-15733.	-3464.	-4611.	505.5	109499.
	31.8	0.0	44.2	15795.	-2425.	-80.	-15733.	-3464.	-4611.	740.	107708.
	0.148		76.9	17034.	-2615.	-86.	-16967.	-3734.	-4973.	0.230	1950.
	95.	0.2204	49.9		-76.18						2034.
	684.2	0.9272	46.9	2.7413	-0.2209	-0.0139	-0.5196	-0.0197	-0.0244		0.01320
18	100.4	0.0	45.0	17110.	-2196.	-69.	-17409.	-3521.	-5264.	505.7	109537.
	31.6	0.0	44.3	17110.	-2196.	-69.	-17409.	-3521.	-5264.	740.	102118.
	0.147		76.5	18451.	-2360.	-74.	-18774.	-3180.	-5677.	0.229	1962.4
	95.	0.2204	50.1		-64.33						2051.5
	684.2	0.9273	47.0	2.9473	-0.3631	-0.0120	-0.5764	-0.0191	-0.0296		0.013363
19	100.5	0.0	45.0	18230.	-1913.	-87.	-18153.	-3530.	-5739.	505.6	110615.
	31.7	0.0	44.5	18230.	-1913.	-87.	-18153.	-3530.	-5739.	740.	102372.
	0.147		76.5	19660.	-2063.	-94.	-19576.	-3803.	-6189.	0.229	1913.3
	95.	0.2204	49.4		-66.32						2103.5
	684.2	0.9273	47.7	3.1757	-0.3332	-0.0152	-0.6018	-0.0195	-0.0311		0.013463
20	100.8	0.0	45.0	18350.	-1907.	-93.	-18215.	-3441.	-5488.	505.5	110696.
	31.9	0.0	44.4	18350.	-1907.	-93.	-18215.	-3441.	-5488.	740.	103254.
	0.148		76.9	19790.	-2057.	-100.	-19645.	-3711.	-5919.	0.230	1921.6
	95.	0.2204	50.3		-59.82						2072.6
	684.2	0.9272	47.3	3.1900	-0.3305	-0.0161	-0.6007	-0.0185	-0.0296		0.013515
21	100.7	0.0	45.0	18865.	-1771.	-74.	-18502.	-3306.	-5557.	505.7	111575.
	31.8	0.0	44.4	18865.	-1771.	-74.	-18502.	-3306.	-5557.	741.	103367.
	0.147		76.8	20403.	-1915.	-80.	-19989.	-3574.	-6004.	0.230	1924.3
	96.	0.2200	50.0		-55.72						2088.
	684.8	0.9256	47.0	3.2829	-0.3078	-0.0129	-0.6120	-0.0179	-0.0300		0.013574
22	100.7	0.0	45.0	18330.	-1922.	-67.	-18106.	-1890.	2323.	505.7	110302.
	31.8	0.0	44.6	18330.	-1922.	-67.	-18106.	-1890.	2323.	740.	109942.
	0.147		77.8	19804.	-2077.	-72.	-19562.	-2042.	2510.	0.229	1932.6
	96.	0.2230	59.5		-60.51						2088.2
	684.8	0.9256	47.0	3.1884	-0.3343	-0.0117	-0.5993	-0.0102	0.0126		0.013603
23	100.7	0.0	45.0	18260.	-1935.	14.	-17791.	562.	14682.	505.7	98432.
	31.7	0.0	44.5	18260.	-1935.	14.	-17791.	562.	14682.	740.	110365.
	0.147		77.6	19763.	-2094.	15.	-19255.	630.	15890.	0.229	1927.9
	97.	0.2196	70.0		-61.66						2060.5
	685.4	0.9240	47.0	3.1833	-0.3373	0.0024	-0.5902	0.0032	0.0796		0.013591
24	100.7	0.0	45.0	18100.	-1896.	32.	-18069.	2050.	21619.	505.5	92540.
	31.7	0.0	44.6	18100.	-1896.	32.	-18069.	2050.	21619.	740.	121164.
	0.147		77.7	19590.	-2052.	35.	-19492.	2218.	23399.	0.230	1416.3
	97.	0.2196	74.3		-59.70						2074.0
	685.4	0.9239	46.4	3.1567	-0.3300	0.0056	-0.5965	0.0111	0.1170		0.013537
25	100.7	0.0	45.0	18510.	-1875.	-68.	-12783.	-2601.	-3483.	505.5	110256.
	31.7	0.0	44.6	18510.	-1875.	-68.	-12783.	-2601.	-3483.	740.	104631.
	0.147		68.8	20034.	-2024.	-74.	-13836.	-2615.	-3770.	0.230	1924.9
	97.	0.2196	50.1		-59.12						2080.7
	685.4	0.9239	46.9	3.2285	-0.3266	-0.0116	-0.4238	-0.0141	-0.0189		0.013614
26	100.8	0.0	45.0	18080.	-1950.	-111.	-23044.	-2690.	-3756.	506.0	109757.
	31.8	0.0	44.6	18080.	-1950.	-111.	-23044.	-2690.	-3756.	741.	104424.
	0.147		69.3	19569.	-2111.	-120.	-24943.	-2912.	-4066.	0.230	1427.4
	97.	0.2196	51.7		-61.35						2080.7
	685.4	0.9239	47.0	3.1825	-0.3389	-0.0193	-0.7624	-0.0145	-0.0203		0.013572
27	101.[illegible]	0.0	45.0	18510.	-1869.	-326.	-18488.	2350.	-6408.	505.6	110530.
	32.0	0.0	63.1	18510.	-1869.	-326.	-18488.	2350.	-6408.	740.	103504.
	0.148		79.1	19819.	-2022.	-353.	-20012.	2544.	-6936.	0.210	1920.9
	97.	0.2196	50.0		-56.46						2079.2
	685.4	0.9239	47.0	3.1668	-0.3230	-0.0564	-0.6083	0.0126	-0.0344		0.013547
28	100.9	0.0	45.0	18295.	-1846.	-575.	-18201.	8412.	-7008.	505.7	110576.
	31.8	0.0	53.9	18295.	-1846.	-575.	-18201.	8412.	-7008.	741.	103523.
	0.147		76.9	19857.	-2004.	-623.	-19735.	9121.	-7599.	0.230	1921.5
	98.	0.2192	50.7		-58.14						2083.6
	686.0	0.9222	46.4	3.1799	-0.3212	-0.0999	-0.6020	0.0455	-0.0379		0.013564
29	80.8	+5.0	80.0	9323.	+99.	+55.	+782.	-2035.	-1448.	506.0	47318.
	20.5	0.0	44.2	9320.	+99.	+55.	+782.	-2035.	-1448.	741.	45262.
	0.118		65.6	10046.	+107.	+59.	+843.	-2194.	-1561.	0.188	831.4
	97.	0.2205	50.7		+4.53						898.4
	685.4	0.9275	19.1	2.5335	+0.0207	+0.0148	+0.0461	-0.0171	-0.0121		0.005824
30	80.5	-3.0	83.0	10545.	61.	+76.	-2504.	-1667.	[illegible]	505.9	46599.
	20.4	0.0	44.6	10545.	61.	+76.	-2504.	-1667.	[illegible]	742.	44599.
	0.118		65.5	11364.	66.	+82.	-2700.	-1797.	[illegible]	0.184	821.5
	97.	0.2205	50.7		3.00						895.6
	685.4	0.9276	19.2	2.4607	0.0165	-0.0226	-0.1293	-0.0141	[illegible]		0.005763
31	80.6	-1.0	80.0	11880.	224.	+45.	-3317.	-1687.	-1573.	505.9	47258.
	20.4	0.0	44.7	11880.	224.	+45.	-3317.	-1687.	-1573.	741.	44458.
	0.118		65.5	12800.	241.	+49.	-3576.	-2035.	-2014.	0.184	827.8
	97.	0.2205	50.[illegible]		10.47						892.5
	685.4	0.9275	19.2	3.2153	0.0606	+0.0122	-0.1706	-0.0159	-0.0156		0.0058[illegible]
32	80.4	[illegible]	80.0	13130.	455.	-22.	-3069.	-1953.	-837.	505.8	44814.
	20.3	0.0	33.7	13130.	455.	-22.	-3069.	-1953.	-837.	741.	44374.
	0.118		65.6	14155.	491.	-24.	-3477.	-2102.	-902.	0.185	820.5
	97.	0.2205	50.7		22.59						884.6
	685.4	0.9276	19.2	3.5734	0.1237	-0.0[illegible]	-0.[illegible]	-0.0165	-0.0[illegible]		0.005755

TABLE V.— Continued.

RUN 26 TEST 525

PT	VKTS / QPSF / MACH / TEMP / VSNDKTS	ALPS.C / PSI / / PHUGDEG / PHDF	INA / PEDRGS / LONG STK / LAT STK / PORTCET	LIFT.G / LIFT / LIFT/Q / / CL	DRAG.G / DRAG / DRAG/Q / FE / CD	SIDE.G / SIDE / SIDE/Q / / CY	PITCH.G / PITCH / PITCH/Q / / CM	YAW.G / YAW / YAW/Q / / CN	ROLL.G / ROLL / ROLL/Q / / CROLL	RPM / OMEGA / V/OR	CMAS / CMAS / HP / HP/Q / CP/SIGMA
33	80.4	3.0	80.0	14590.	757.	-56.	-3707.	-1469.	-652.	565.8	46679.
	20.3	3.0	44.5	14590.	757.	-56.	-3707.	-1469.	-652.	741.	44833.
	0.118		65.8	15632.	810.	-60.	-3944.	-1564.	-702.	0.163	623.3
	97.	0.2205	56.7		37.30						637.5
	685.4	0.9276	19.2	3.9478	0.2060	-0.0152	-0.1442	-0.0124	-0.0055		0.005779
34	80.7	-1.0	80.0	12790.	463.	-65.	6081.	-1417.	-1391.	565.9	47478.
	20.5	0.0	44.5	12790.	463.	-65.	6081.	-1417.	-1391.	741.	45516.
	0.118		55.9	13784.	499.	-70.	6554.	-1528.	-1284.	0.164	633.0
	97.	0.2205	50.2		22.65						400.2
	685.4	0.9275	19.2	3.0534	0.1250	-0.0176	0.1375	-0.0114	-0.0100		0.005857
35	80.5	-1.0	80.0	11230.	43.	-55.	-6913.	-1837.	-1037.	566.0	47776.
	20.4	0.0	44.7	11230.	63.	-55.	-6913.	-1837.	-1037.	741.	45555.
	0.118		75.1	12107.	46.	-59.	-7454.	-1981.	-1118.	0.163	638.1
	97.	0.2205	50.0		2.31						403.6
	685.4	0.9276	19.2	3.0465	0.0117	-0.0149	-0.1860	-0.0155	-0.0067		0.005876
36	80.6	-1.0	80.0	11740.	284.	8.	-3306.	-1562.	23056.	565.8	34984.
	20.4	0.0	44.6	11740.	266.	8.	-3306.	-1562.	23056.	741.	52616.
	0.118		64.0	12657.	3[illegible].	9.	-3242.	-2034.	24857.	0.164	624.4
	97.	0.2205	50.0		14.11						608.8
	685.4	0.9275	19.2	3.1775	0.0770	0.0022	-0.1549	-0.0159	0.1440		0.005794
37	80.0	-1.0	80.0	11675.	300.	-15.	-3061.	-1744.	23007.	566.2	38024.
	20.4	0.0	44.7	11675.	300.	-15.	-3061.	-1744.	23007.	741.	52581.
	0.118		64.9	12587.	330.	-16.	-3350.	-1885.	25510.	0.164	620.0
	97.	0.2205	70.0		14.49						605.0
	685.4	0.9275	19.2	3.1598	0.0828	-0.0041	-0.1576	-0.0147	0.1491		0.005756
38	80.4	-1.0	80.0	11020.	403.	30.	-2101.	-4907.	45226.	566.	33274.
	20.3	0.0	44.7	11020.	403.	30.	-2101.	-4907.	45226.	741.	50464.
	0.118		64.7	11880.	434.	32.	-2265.	-4978.	48794.	0.163	610.4
	97.	0.2205	64.6		14.84						673.0
	685.4	0.9276	19.2	2.9968	0.1096	0.0082	-0.1247	-0.0107	0.3821		0.005674
39	80.5	-1.0	80.0	11865.	271.	-237.	-2514.	4200.	-7606.	566.1	40500.
	20.6	0.0	44.3	11865.	271.	-237.	-2514.	4200.	-7606.	741.	45399.
	0.118		64.8	12784.	292.	-255.	-2709.	4521.	-7541.	0.164	624.2
	96.	0.2209	50.0		13.18						627.0
	684.8	0.9292	19.2	3.1887	0.0728	-0.0637	-0.1286	0.0351	-0.0565		0.005765
40	80.5	-1.0	80.0	11860.	268.	-342.	-2396.	4482.	-13991.	566.	46100.
	20.4	0.0	43.2	11860.	268.	-342.	-2396.	4482.	-13991.	741.	45965.
	0.118		64.9	12783.	311.	-369.	-2582.	4842.	-15057.	0.163	627.4
	96.	0.2209	50.1		14.19						690.4
	684.8	0.9292	19.2	3.2174	0.0721	-0.0928	-0.1239	0.0842	-0.1186		0.005768
41	80.5	-1.0	80.0	11830.	275.	-585.	-2797.	-85.	-3309.	566.0	46049.
	20.4	-4.0	43.6	11830.	275.	-585.	-2797.	-85.	-3309.	741.	44949.
	0.118		64.9	12731.	296.	-624.	-3016.	-92.	-3561.	0.163	625.1
	96.	0.2209	50.1		13.44						607.4
	684.8	0.9292	19.2	3.2055	0.0745	-0.1572	-0.1442	-0.0007	-0.0274		0.005772
42	80.6	-1.0	80.0	11825.	254.	-287.	-2466.	-1466.	-696.	566.3	46754.
	20.4	-2.0	43.7	11825.	254.	-287.	-2466.	-1466.	-696.	741.	44716.
	0.118		64.9	12726.	274.	-309.	-2653.	-1578.	-752.	0.164	622.9
	96.	0.2209	50.0		12.67						604.5
	684.8	0.9292	19.2	3.1966	0.0703	-0.0776	-0.1268	-0.0123	-0.0059		0.005743
43	80.6	-1.0	80.0	11825.	250.	211.	-2526.	-2607.	-571.	566.1	44659.
	20.4	2.0	43.5	11825.	250.	211.	-2526.	-2607.	-571.	741.	44716.
	0.118		64.9	12720.	269.	227.	-2719.	-2806.	-614.	0.163	620.6
	96.	0.2209	50.2		12.25						603.1
	684.8	0.9292	19.2	3.2004	0.0677	0.0571	-0.1301	-0.0217	-0.0048		0.005738
44	80.6	-1.0	80.0	11910.	253.	498.	-2340.	-3493.	1714.	566.0	47199.
	20.8	4.0	43.6	11910.	253.	498.	-2340.	-3493.	1714.	741.	45282.
	0.118		64.8	12840.	273.	526.	-2523.	-3766.	1848.	0.164	650.5
	97.	0.2205	44.4		12.39						495.4
	685.4	0.9275	19.2	3.2234	0.0685	0.1321	-0.1205	-0.0294	0.0144		0.005823
45	80.7	-1.0	80.0	11925.	307.	1082.	-999.	-7120.	3105.	566.0	47618.
	20.5	8.0	43.6	11925.	307.	1082.	-999.	-7120.	3105.	741.	45535.
	0.118		64.5	12857.	33[illegible].	1167.	-1077.	-7684.	3350.	0.164	636.6
	97.	0.2205	50.2		15.00						492.0
	685.4	0.9275	19.2	3.2199	0.0829	0.2922	-0.0513	-0.0596	0.0261		0.005864
46	80.6	-1.0	80.0	11665.	410.	1628.	360.	-9216.	6200.	566.0	47074.
	20.4	12.0	43.7	11665.	410.	1628.	360.	-9216.	6200.	741.	44464.
	0.118		64.9	12576.	442.	1755.	385.	-9936.	6685.	0.164	626.7
	97.	0.2205	50.1		22.40						641.2
	685.4	0.9275	19.2	3.1577	0.1110	0.4406	0.0165	-0.0776	0.0522		0.005795

TABLE V.— Continued.

RUN 29 TEST 525

PT	VKTS / QPSF / MACH / TEMP / [illegible]	ALPSTC / PSI / / [illegible] / [illegible]	INB PRO POS / CONV STK / LAT STK / [illegible]	LIFT,LB / LIFT / LIFT/Q / CL	DRAG,LB / DRAG / DRAG/Q / FE / CD	SIDE,LB / SIDE / SIDE/Q / CY	PITCH,LB / PITCH / PITCH/Q / CM	YAW,LB / YAW / YAW/Q / CN	ROLL,LB / ROLL / ROLL/Q / CROLL
3	119.1	-4.0	60.0	4755.	2890.	-155.	16037.	-91.	598.
	49.8	0.0	49.9	4755.	2890.	-155.	16037.	-91.	598.
	0.177		50.7	4939.	3027.	-162.	16795.	-95.	724.
	79.	0.2270	49.7		63.07				
	675.0	0.9548	50.0	0.7665	0.3868	-0.0127	0.3679	-0.0023	0.0025
4	119.1	-2.0	60.0	5590.	2433.	-155.	15744.	-540.	-366.
	49.7	0.0	49.9	5590.	2433.	-155.	15744.	-540.	-366.
	0.177		50.7	6165.	3027.	-155.	17579.	-567.	-384.
	79.	0.2265	49.9		64.13				
	678.2	0.9537	49.9	0.7118	0.3593	-0.0191	0.3878	-0.0020	-0.0014
5	118.4	0.0	60.0	7040.	3043.	-140.	14182.	-303.	-1359.
	49.2	0.0	49.8	7048.	3083.	-140.	14182.	-303.	-1359.
	0.176		50.7	7438.	3192.	-147.	14875.	-432.	-1405.
	79.	0.2266	49.9		67.29				
	678.2	0.9537	50.0	0.8682	0.3717	-0.0171	0.3297	-0.0015	-0.0051
6	119.3	2.0	60.0	8344.	3166.	-121.	13520.	-441.	-440.
	49.1	0.0	49.9	8344.	3166.	-121.	13520.	-441.	-440.
	0.176		50.7	8765.	3350.	-127.	14107.	-460.	-452.
	80.	0.2262	50.0		70.67				
	679.4	0.9515	50.1	1.0214	0.3909	-0.0148	0.2685	-0.0017	-0.0017
7	119.0	4.0	60.0	9442.	3354.	-118.	9162.	-105.	105.
	49.4	0.0	49.7	9442.	3354.	-118.	9162.	-105.	105.
	0.176		50.7	9922.	3525.	-124.	9625.	-113.	111.
	80.	0.2262	50.0		74.74				
	679.4	0.9510	50.0	1.1621	0.4128	-0.0145	0.2146	-0.0004	0.0004
8	118.3	5.0	60.0	10150.	3460.	-120.	7508.	373.	31.
	49.0	0.0	49.9	10150.	3460.	-120.	7508.	373.	31.
	0.176		50.7	10667.	3645.	-126.	7965.	392.	32.
	81.	0.2256	50.0		76.90				
	679.3	0.9498	50.0	1.2461	0.4249	-0.0147	0.1771	0.0014	0.0001
9	118.4	6.0	60.0	10735.	3555.	-97.	6325.	370.	-64.
	49.1	0.0	49.9	10735.	3555.	-97.	6325.	370.	-64.
	0.176		50.7	11262.	3743.	-102.	6660.	398.	-71.
	81.	0.2258	49.9		76.54				
	675.5	0.9494	49.9	1.3137	0.4339	-0.0119	0.1476	0.0014	-0.0003
10	119.7	-4.0	45.0	5105.	2563.	-189.	15594.	153.	351.
	48.0	0.0	49.8	5105.	2563.	-189.	15594.	153.	351.
	0.176		50.7	5387.	2704.	-199.	16454.	161.	371.
	82.	0.2255	50.0		55.73				
	676.1	0.9477	50.0	0.6133	0.3079	-0.0227	0.3565	0.0006	0.0013
11	119.6	-4.0	45.0	6545.	2641.	-117.	15621.	-135.	744.
	49.4	0.0	49.9	6545.	2641.	-117.	15621.	-135.	744.
	0.176		50.7	6906.	2787.	-123.	16493.	-142.	790.
	82.	0.2253	50.0		57.52				
	676.1	0.9477	50.0	0.7875	0.3178	-0.0141	0.3622	-0.0005	0.0028
12	119.5	0.0	45.0	7770.	2755.	-160.	13656.	-124.	337.
	49.8	0.0	50.0	7770.	2755.	-160.	13656.	-124.	337.
	0.177		50.7	8194.	2907.	-169.	14406.	-133.	356.
	82.	0.2253	49.9		60.15				
	675.1	0.9478	50.0	0.9373	0.3323	-0.0193	0.3135	-0.0005	0.0013
13	119.4	2.0	45.0	9000.	2916.	-143.	11294.	199.	-310.
	49.8	0.0	50.0	9000.	2916.	-143.	11294.	199.	-310.
	0.177		50.7	9513.	3082.	-151.	11937.	211.	-328.
	83.	0.2249	50.0		63.69				
	675.7	0.9481	50.0	1.0895	0.3530	-0.0173	0.2602	0.0007	-0.0012
14	119.4	4.0	45.0	10225.	3111.	-141.	8655.	261.	-202.
	49.7	0.0	49.8	10225.	3111.	-141.	8655.	261.	-202.
	0.177		50.7	10808.	3288.	-149.	9148.	276.	-214.
	83.	0.2249	50.0		68.13				
	676.7	0.9481	50.0	1.2371	0.3764	-0.0171	0.2039	0.0010	-0.0008
15	119.1	6.0	45.0	11470.	3290.	-131.	6327.	845.	105.
	49.8	0.0	50.0	11470.	3290.	-131.	6327.	845.	105.
	0.177		50.7	12123.	3477.	-138.	6687.	893.	111.
	83.	0.2249	50.0		72.45				
	676.7	0.9482	50.0	1.3955	0.4003	-0.0159	0.1465	0.0032	0.0004
16	119.0	6.0	45.0	12620.	3554.	-137.	3382.	847.	226.
	49.4	0.0	49.1	12620.	3554.	-137.	3382.	847.	226.
	0.177		50.6	13338.	3756.	-145.	3573.	896.	239.
	85.	0.2249	50.0		76.31				
	678.7	0.9482	50.0	1.5382	0.4378	-0.0167	0.0783	0.0032	0.0009
17	119.0	8.0	45.0	12380.	3539.	-182.	4230.	262.	5122.
	49.4	0.0	50.0	12380.	3539.	-182.	4230.	262.	5122.
	0.176		50.6	13084.	3740.	-192.	4471.	277.	5413.
	83.	0.2249	70.0		78.32				
	678.7	0.9482	50.0	1.5074	0.4310	-0.0222	0.0980	0.0010	0.0194
18	119.0	8.0	45.0	11955.	3448.	-155.	6348.	-462.	10324.
	49.3	0.0	49.9	11955.	3448.	-155.	6348.	-462.	10324.
	0.176		50.7	12635.	3642.	-164.	6707.	-1017.	10912.
	83.	0.2249	49.5		76.[illegible]				
	675.7	0.9482	49.9	1.4580	0.4199	-0.0189	0.1472	-0.0036	0.0390
19	121.6	-4.0	[illegible]	5405.	1590.	-90.	12187.	924.	1494.
	47.9	0.0	49.9	5405.	1590.	-90.	12187.	924.	1494.
	0.180		50.7	5727.	1685.	-95.	12915.	979.	1583.
	84.	0.2243	49.9		33.03				
	677.4	0.9435	50.0	0.6517	0.1958	-0.0105	0.2731	0.0034	0.0054
20	122.1	-2.0	[illegible]	7500.	1615.	-113.	12460.	674.	573.
	47.8	0.0	50.0	7500.	1615.	-113.	12460.	674.	573.
	0.181		50.7	7943.	1714.	-124.	13258.	710.	602.
	85.	0.2234	50.0		33.45				
	678.0	0.9424	50.0	0.8124	0.1876	-0.0126	0.2584	0.0024	0.0022

TABLE V.— Continued.

RUN 29 TEST 525

PT	KNOTS QPSF MACH TEMP VSPDKTS	ALFS,C PSI HUMIDITY RHUP	INA PED POS LONG STK LAT STK POR CTR	LIFT LIFT LIFT/Q CL	DRAG DRAG DRAG/Q CD	SIDE SIDE SIDE/Q CY	PITCH PITCH PITCH/Q CM	YAW YAW YAW/Q CN	ROLL ROLL ROLL/Q CROLL
21	122.0	6.0	0.0	8542.	1646.	-75.	10566.	1104.	1232.
	47.4	0.0	50.2	8542.	1646.	-75.	10566.	1168.	1232.
	0.161		50.7	9066.	1744.	-80.	11237.	1240.	1306.
	85.	0.2234	50.0		34.74				
	678.0	0.9820	50.0	0.9907	0.1914	-0.0087	0.2346	0.0642	0.0045
22	121.7	2.0	0.0	10220.	1771.	-114.	6364.	366.	911.
	47.3	0.0	50.0	10220.	1771.	-114.	6364.	366.	911.
	0.160		50.7	10636.	1850.	-120.	6960.	1452.	967.
	85.	0.2239	50.0		37.48				
	678.0	0.9821	49.9	1.1716	0.2071	-0.0139	0.1865	0.0050	0.0033
23	121.5	4.0	0.0	11480.	1910.	-130.	5496.	1520.	952.
	47.3	0.0	50.0	11080.	1910.	-130.	5496.	1523.	952.
	0.160		50.7	12185.	2022.	-135.	5834.	1613.	1011.
	85.	0.2239	50.0		40.56				
	678.0	0.9821	50.0	1.3489	0.2291	-0.0153	0.1227	0.0055	0.0035
24	121.5	6.0	0.0	13050.	2086.	-91.	2963.	1276.	754.
	47.3	0.0	50.0	13050.	2086.	-91.	2963.	1276.	754.
	0.160		50.7	13852.	2214.	-101.	3150.	1357.	800.
	85.	0.2239	50.0		44.30				
	678.0	0.9821	50.0	1.5311	0.2097	-0.0110	0.0662	0.0047	0.0027
25	121.3	6.0	0.0	14485.	2280.	-117.	-51.	1536.	1340.
	46.9	0.0	50.0	14485.	2280.	-117.	-51.	1536.	1340.
	0.160		50.7	15374.	2425.	-124.	-55.	1631.	1422.
	85.	0.2239	50.0		44.71				
	678.0	0.9822	49.9	1.7053	0.2691	-0.0138	-0.0012	0.0056	0.0049
26	121.5	10.0	0.0	15895.	2528.	-126.	-3380.	1456.	652.
	47.0	0.0	50.0	15895.	2528.	-126.	-3380.	1456.	652.
	0.160		50.6	16902.	2689.	-134.	-3594.	1549.	693.
	86.	0.2235	50.0		53.74				
	678.6	0.9806	49.9	1.8685	0.2972	-0.0148	-0.0756	0.0053	0.0024
27	121.2	11.0	0.0	16530.	2644.	-118.	-3699.	1201.	639.
	46.7	0.0	50.0	16530.	2644.	-118.	-3699.	1201.	639.
	0.179		50.7	17575.	2811.	-125.	-3933.	1277.	679.
	86.	0.2236	50.0		56.56				
	679.0	0.9805	49.9	1.9537	0.3125	-0.0139	-0.0832	0.0044	0.0023
28	121.3	8.0	0.0	14120.	2245.	-206.	420.	522.	5954.
	46.7	0.0	50.0	14120.	2245.	-206.	420.	522.	5954.
	0.179		50.5	15044.	2391.	-219.	446.	556.	6342.
	87.	0.2231	49.2		48.03				
	679.2	0.9384	49.8	1.6676	0.2652	-0.0243	0.0095	0.0019	0.0214
29	121.2	6.0	0.0	13515.	2166.	-252.	2766.	12.	13210.
	46.7	0.0	50.5	13515.	2166.	-252.	2766.	12.	13210.
	0.179		50.5	14396.	2328.	-274.	2964.	13.	14071.
	87.	0.2231	49.6		46.79				
	679.2	0.9384	49.7	1.5981	0.2585	-0.0310	0.0627	0.0000	0.0466
30	122.5	-4.0	0.0	3335.	1254.	-60.	13765.	1771.	3599.
	47.7	0.0	50.0	3335.	1254.	-60.	13765.	1771.	3599.
	0.161		50.7	3554.	1336.	-65.	14667.	1887.	4155.
	87.	0.2231	50.0		26.31				
	679.2	0.9385	49.8	0.3866	0.1458	-0.0093	0.3036	0.0066	0.0140
31	122.5	-2.0	0.0	5020.	1239.	-72.	12069.	1858.	2528.
	47.7	0.0	50.0	5020.	1239.	-72.	12069.	1858.	2528.
	0.161		50.7	5349.	1320.	-77.	13733.	1980.	2693.
	87.	0.2231	50.0		25.97				
	679.2	0.9385	49.8	0.5813	0.1435	-0.0083	0.2641	0.0067	0.0091
32	122.5	0.0	0.0	6525.	1276.	-84.	11763.	1759.	2129.
	47.7	0.0	50.0	6525.	1276.	-84.	11763.	1759.	2129.
	0.181		50.6	6955.	1362.	-90.	12534.	1874.	2269.
	87.	0.2231	50.0		26.81				
	679.2	0.9385	49.8	0.7559	0.1481	-0.0097	0.2595	0.0063	0.0077
33	122.3	2.0	0.0	8050.	1357.	-63.	8633.	1562.	2303.
	47.5	0.0	50.6	8050.	1357.	-63.	8633.	1562.	2303.
	0.181		50.6	8577.	1446.	-67.	9196.	1664.	2454.
	87.	0.2231	50.0		28.55				
	679.2	0.9385	49.8	0.9356	0.1577	-0.0073	0.1910	0.0056	0.0083
34	122.1	4.0	0.0	9500.	1488.	-104.	6697.	2020.	1901.
	47.4	0.0	50.5	9500.	1488.	-104.	6697.	2021.	1901.
	0.180		50.6	10122.	1585.	-111.	7134.	2152.	2025.
	87.	0.2231	50.0		31.4.				
	679.2	0.9386	49.8	1.1047	0.1730	-0.0121	0.1467	0.0075	0.0064
35	122.0	6.0	0.0	11025.	1634.	-114.	3645.	1957.	3055.
	47.3	0.0	50.5	11025.	1634.	-114.	3645.	1957.	3055.
	0.180		50.7	11746.	1741.	-121.	4054.	2085.	3255.
	87.	0.2231	50.0		34.55				
	679.2	0.9385	49.7	1.2872	0.1908	-0.0133	0.0854	0.0071	0.0111
36	122.0	8.0	0.0	12395.	1643.	-105.	1265.	1706.	2378.
	47.2	0.0	50.0	12395.	1643.	-105.	1265.	1706.	2378.
	0.180		50.6	13224.	1967.	-112.	1372.	1823.	2535.
	88.	0.2227	50.0		35.04				
	679.9	0.9369	49.7	1.4507	0.2157	-0.0123	0.0248	0.0062	0.0086
37	121.7	10.0	0.0	13950.	2040.	-93.	-1088.	1228.	1826.
	47.0	0.0	50.7	13950.	2040.	-93.	-1088.	1228.	1826.
	0.180		50.7	14863.	2176.	-99.	-1152.	1310.	1946.
	89.	0.2226	50.0		43.43				
	679.9	0.9373	49.8	1.6407	0.2399	-0.0109	-0.0242	0.0045	0.0067
38	122.3	-4.0	0.0	405.	1213.	5.	11573.	2055.	4044.
	47.5	0.0	50.6	405.	1213.	5.	11573.	2055.	4044.
	0.181		50.7	432.	1294.	5.	12265.	2193.	4315.
	88.	0.2229	50.0		25.56				
	679.9	0.9372	49.7	0.0471	0.1412	0.0006	0.2550	0.0074	0.0146

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TABLE V.— Continued.

RUN 29 TEST 525

PT	VATS / QPSF / MION / TEMP / VSNDKTS	ALFS.C / PSI / / RHO*ICC / RHOR	INA / PET POS / LONG STK / LAT STK / POW LEV	LIFT.C / LIFT / LIFT/A / / CL	DRAG.C / DRAG / DRAG/A / RE / CD	SIDE.C / SIDE / SIDE/A / / CY	PITCH.C / PITCH / PITCH/A / / CM	YAW.C / YAW / YAW/A / / CN	ROLL.C / ROLL / ROLL/A / / CROLL
39	122.5	-2.0	0.0	2601.	1157.	-3.	5947.	2506.	3413.
	47.6	0.0	50.?	2509.	1157.	-3.	4947.	2506.	3413.
	0.181		50.7	2734.	1235.	-3.	10561.	2674.	4176.
	88.	0.2227	50.0		24.50				
	679.9	0.9377	49.7	0.2320	0.1142	+0.0003	0.2185	0.0040	0.0141
40	122.6	0.0	0.0	3651.	1125.	-26.	8264.	2285.	3057.
	47.8	0.0	50.5	3630.	1125.	-29.	8264.	2285.	3057.
	0.181		50.6	3881.	1203.	-21.	8834.	2442.	3269.
	89.	0.2223	50.1		25.62				
	680.5	0.9355	49.7	0.4212	0.1305	+0.0023	0.1824	0.0042	0.0110
41	122.6	2.0	0.0	5105.	1134.	-37.	5683.	2526.	3725.
	47.8	0.0	50.3	5105.	1134.	-37.	5683.	2526.	3725.
	0.181		50.6	5457.	1216.	-40.	6076.	2702.	3982.
	89.	0.2223	50.1		25.44				
	680.5	0.9354	49.7	0.5929	0.1323	+0.0043	0.1255	0.0041	0.0134
42	122.4	4.0	0.0	6670.	1190.	-20.	3153.	1970.	3373.
	47.5	0.0	50.5	6670.	1190.	-20.	3153.	1970.	3373.
	0.181		50.6	7134.	1272.	-21.	3371.	2106.	3605.
	89.	0.2223	50.0		25.60				
	680.5	0.9355	49.7	0.7781	0.1385	+0.0023	0.0698	0.0071	0.0122
43	122.3	6.0	0.0	8110.	1296.	+25.	95.	2169.	3938.
	47.3	0.0	50.6	8110.	1296.	+25.	95.	2169.	3938.
	0.180		50.7	8665.	1385.	+27.	1015.	2318.	4103.
	89.	0.2224	50.0		27.37				
	680.5	0.9355	49.7	0.9466	0.1512	-0.0029	0.0211	0.0074	0.0139
44	122.0	8.0	0.0	9405.	1431.	2.	-4424.	1702.	4165.
	47.1	0.0	50.5	9405.	1431.	2.	-4424.	1702.	4165.
	0.180		50.7	10055.	1530.	2.	-4866.	1819.	4451.
	89.	0.2224	50.0		30.30				
	680.5	0.9354	49.7	1.1022	0.1677	0.0002	-0.0195	0.0062	0.0152
45	121.8	10.0	0.0	10865.	1616.	-1.	-2200.	1422.	3197.
	47.0	0.0	50.8	10865.	1616.	-1.	-2200.	1422.	3197.
	0.180		50.6	11614.	1727.	-1.	-2352.	1520.	3417.
	89.	0.2224	50.0		34.37				
	680.5	0.9356	49.7	1.2769	0.1899	-0.0001	-0.0092	0.0052	0.0117

RUN 30 TEST 525

PT	VATS / QPSF / MION / TEMP / VSNDKTS	ALFS.C / PSI / / RHO*ICC / RHOR	INA / PET POS / LONG STK / LAT STK / POW LEV	LIFT.C / LIFT / LIFT/A / / CL	DRAG.C / DRAG / DRAG/A / RE / CD	SIDE.C / SIDE / SIDE/A / / CY	PITCH.C / PITCH / PITCH/A / / CM	YAW.C / YAW / YAW/A / / CN	ROLL.C / ROLL / ROLL/A / / CROLL
3	118.7	-4.0	0.0	4595.	1268.	-117.	11955.	1604.	1645.
	45.2	0.0	50.1	4595.	1268.	-117.	11955.	1604.	1645.
	0.176		50.0	4902.	1338.	-125.	12616.	1693.	1759.
	82.	0.2252	50.0		26.05				
	676.1	0.9477	50.2	0.5677	0.1550	-0.0143	0.2761	0.0061	0.0063
4	118.3	-2.0	0.0	6245.	1282.	-121.	11344.	1126.	1580.
	44.9	0.0	50.1	6245.	1282.	-121.	11344.	1126.	1580.
	0.176		50.0	6591.	1353.	-128.	11973.	1190.	1667.
	82.	0.2252	50.0		26.59				
	676.1	0.9474	50.1	0.7692	0.1579	-0.0149	0.2659	0.0043	0.0061
5	118.2	0.0	0.0	7885.	1333.	-134.	10075.	1294.	1716.
	44.8	0.0	50.2	7885.	1333.	-134.	10075.	1294.	1716.
	0.175		50.0	8322.	1407.	-141.	10634.	1366.	1811.
	82.	0.2252	50.0		29.76				
	676.1	0.9475	50.1	0.9727	0.1644	-0.0165	0.2365	0.0050	0.0066
6	118.4	2.0	0.0	9475.	1457.	-115.	8074.	1213.	1608.
	44.9	0.0	50.1	9475.	1457.	-115.	8074.	1213.	1608.
	0.176		50.0	10019.	1541.	-122.	8538.	1282.	1700.
	83.	0.2248	49.9		32.48				
	676.7	0.9457	50.0	1.1671	0.1795	-0.0142	0.1893	0.0046	0.0062
7	118.7	4.0	0.0	11045.	1610.	-124.	5679.	1641.	1247.
	45.0	0.0	50.0	11045.	1610.	-124.	5679.	1641.	1247.
	0.176		50.0	11754.	1706.	-131.	6229.	1734.	1376.
	84.	0.2244	50.0		35.78				
	677.4	0.9439	50.0	1.3624	0.1977	-0.0152	0.1374	0.0063	0.0051
8	118.6	6.0	0.0	12765.	1773.	-126.	3555.	1419.	1555.
	44.9	0.0	50.2	12765.	1773.	-126.	3555.	1419.	1555.
	0.176		50.0	13466.	1872.	-133.	3754.	1505.	1640.
	84.	0.2244	50.0		39.45				
	677.4	0.9439	50.2	1.5617	0.2179	-0.0155	0.0831	0.0054	0.0059
9	118.4	8.0	0.0	13885.	2008.	-141.	488.	1140.	442.
	44.8	0.0	50.1	13885.	2008.	-141.	488.	1140.	442.
	0.175		50.0	14615.	2127.	-149.	1045.	1208.	468.
	84.	0.2244	50.0		44.74				
	677.4	0.9440	50.0	1.6954	0.2475	-0.0176	0.0231	0.0044	0.0017
10	118.6	10.0	0.0	15175.	2237.	-116.	-1304.	865.	[illegible]
	44.9	0.0	50.2	15175.	2237.	-116.	-1304.	865.	[illegible]
	0.176		50.0	16024.	2376.	-123.	-1378.	914.	44.
	84.	0.2244	50.0		49.07				
	677.4	0.9439	50.1	1.8444	0.2751	-0.0145	-0.0306	0.0033	0.0002
11	118.7	4.0	45.0	4535.	2214.	-117.	25464.	445.	1370.
	45.0	0.0	50.1	4535.	2214.	-116.	25464.	445.	1370.
	0.176		50.0	4815.	2350.	-123.	26943.	457.	1455.
	85.	0.2237	49.9		49.0				
	678.0	0.9422	50.1	0.5555	0.2717	-0.0142	0.5724	0.0067	0.0057

TABLE V.— Continued.

RUN 3[illegible] TEST 5[illegible]

PT	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
12	118.5	-2.0	45.0	5930.	2293.	-165.	15357.	612.	599.
	44.6	0.0	50.1	5930.	2293.	-165.	15357.	612.	599.
	0.175		50.0	6245.	2434.	-173.	16288.	644.	636.
	85.	0.2240	50.0		51.23				
	678.0	0.9423	50.1	0.7531	0.2836	-0.0201	[illegible]	0.0123	0.0023
13	119.1	4.0	45.0	7345.	2444.	-149.	13441.	1044.	711.
	45.2	0.0	50.1	7390.	2444.	-149.	13441.	1044.	711.
	0.176		50.1	7595.	2504.	-158.	14425.	1110.	758.
	88.	0.2235	50.0		54.21				
	678.6	0.9408	50.0	0.8972	0.2445	-0.0142	0.3244	0.0040	0.0027
14	118.2	2.0	45.0	8755.	2601.	-155.	11854.	768.	482.
	44.9	0.0	50.0	8755.	2601.	-155.	11854.	768.	482.
	0.176		50.0	9304.	2766.	-147.	12191.	817.	515.
	86.	0.2235	49.5		57.90				
	678.6	0.9409	50.1	[illegible]	0.3144	-0.0228	0.2727	0.0029	0.0016
15	118.7	4.0	45.0	9850.	2808.	-165.	10150.	744.	203.
	44.8	0.0	50.1	9850.	2808.	-165.	10150.	744.	203.
	0.175		50.0	10492.	2991.	-174.	10611.	790.	217.
	87.	0.2231	49.9		62.73				
	679.2	0.9369	50.1	1.2157	0.3466	-0.0201	0.2384	0.0029	0.0008
16	118.9	6.0	45.0	11090.	2985.	-191.	8297.	311.	1655.
	44.9	0.0	50.2	11090.	2985.	-191.	8297.	311.	1655.
	0.176		50.0	11614.	3104.	-154.	8458.	332.	1763.
	87.	0.2231	50.0		65.41				
	679.2	0.9388	50.1	1.3832	0.3669	-0.0173	0.1941	0.0012	0.0063
17	119.0	8.0	45.0	12290.	3209.	-124.	8465.	704.	911.
	45.0	0.0	49.9	12290.	3209.	-124.	8465.	704.	911.
	0.176		50.0	13092.	3482.	-111.	9102.	819.	971.
	87.	0.2231	50.0		72.69				
	679.2	0.9387	50.0	1.5044	0.4016	-0.0128	0.1137	0.0024	0.0035
18	118.9	8.0	45.0	11975.	3284.	-113.	5504.	369.	8182.
	44.9	0.0	50.0	11975.	3284.	-113.	5504.	369.	8182.
	0.176		49.9	12756.	3477.	-120.	5867.	388.	8716.
	87.	0.2231	69.9		72.69				
	679.2	0.9388	50.1	1.4735	0.4036	-0.0139	0.1240	0.0014	0.0513
19	118.8	8.0	45.0	11275.	3233.	-113.	7515.	-562.	16683.
	44.9	0.0	49.8	11275.	3233.	-113.	7515.	-562.	16683.
	0.176		49.9	12010.	3444.	-120.	8324.	-599.	17771.
	87.	0.2231	64.7		72.08				
	679.2	0.9388	50.0	1.3866	0.3982	-0.0139	0.1832	-0.0022	0.0639
20	119.4	8.0	0.0	13845.	2020.	-177.	1443.	640.	10694.
	45.2	0.0	49.9	13845.	2020.	-177.	1443.	640.	10694.
	0.176		49.9	14503.	2156.	-194.	2474.	698.	11414.
	86.	0.2227	[illegible]		44.69				
	679.9	0.9370	50.0	1.6878	0.2469	-0.0216	0.0452	0.0032	0.0406
21	119.9	8.0	0.0	12080.	1978.	-224.	4024.	345.	19142.
	[illegible]	0.0	50.0	12080.	1978.	-224.	4024.	345.	19142.
	0.175		50.0	13510.	2111.	-239.	4300.	368.	20427.
	88.	0.2227	49.5		44.12				
	679.9	0.9371	50.1	1.5602	0.2438	-0.0276	0.0945	0.0013	0.0733
22	117.1	-4.0	60.0	900.	2359.	-76.	14677.	1333.	1987.
	43.8	0.0	50.2	988.	2359.	-76.	14677.	1333.	1987.
	0.173		49.9	1047.	2521.	-81.	15663.	1425.	2123.
	89.	0.2224	50.9		54.33				
	680.5	0.9359	50.0	0.1207	0.3002	-0.0097	0.3554	0.0653	0.0079
23	119.0	-2.0	60.0	2245.	2482.	-65.	13309.	730.	1765.
	44.8	0.0	50.1	2245.	2482.	-65.	13309.	734.	1765.
	0.175		49.9	2400.	2632.	-91.	14229.	785.	1897.
	89.	0.2223	50.4		58.92				
	682.9	0.9354	50.1	0.2727	0.3039	-0.0105	0.1121	0.0028	[illegible]
24	119.5	0.0	60.0	3810.	2508.	-87.	11216.	752.	1644.
	45.2	0.0	49.9	3810.	2508.	-87.	11216.	752.	1644.
	0.176		50.0	3960.	2682.	-93.	11992.	804.	1758.
	89.	0.2223	50.4		55.43				
	680.5	0.9357	50.0	0.4640	0.3062	-0.0106	0.2808	0.0029	0.0082
25	119.3	2.0	60.0	4605.	2586.	-104.	8983.	1043.	1509.
	44.8	0.0	50.0	4685.	2586.	-104.	8983.	1043.	1509.
	0.175		50.1	5211.	2750.	-111.	9600.	1118.	1402.
	90.	0.2219	50.3		57.31				
	681.1	0.9337	49.9	[illegible]	0.3186	-0.0126	[illegible]	0.0040	0.0050
26	119.1	4.0	60.0	8190.	2654.	-59.	7119.	933.	1124.
	44.9	0.0	50.0	8190.	2654.	-59.	7119.	933.	1124.
	0.176		50.0	8638.	2843.	-70.	7625.	999.	1204.
	90.	0.2219	50.0		59.17				
	681.1	0.9337	50.1	0.7628	0.3245	-0.0065	0.1564	0.0035	0.0043
27	119.2	6.0	60.0	7400.	2809.	-74.	4465.	664.	625.
	44.9	0.0	49.9	7400.	2809.	-74.	4465.	664.	625.
	0.176		50.0	7926.	3009.	-79.	5134.	717.	658.
	90.	0.2219	50.5		62.56				
	681.1	0.9336	50.0	0.9105	0.3456	-0.0091	0.1167	0.0028	0.0024
28	119.2	7.0	60.0	8170.	2870.	-40.	4125.	1271.	635.
	44.9	0.0	49.9	8170.	2870.	-41.	4125.	1271.	635.
	0.175		50.0	8751.	3070.	-43.	4412.	1361.	680.
	90.	0.2219	50.4		65.95				
	681.1	0.9337	50.0	1.0358	0.3553	-0.0101	[illegible]	0.0144	0.0023
29	119.0	9.0	60.0	8725.	2961.	-41.	3157.	438.	1077.
	44.9	0.0	49.9	8725.	2961.	-41.	3157.	438.	1077.
	0.175		50.0	9345.	3150.	-47.	3275.	[illegible]	1201.
	90.	0.2219	50.0		[illegible]				
	681.1	0.9337	50.1	1.0769	0.3650	[illegible]	[illegible]	0.0036	[illegible]
30	119.0	9.0	60.0	9350.	3033.	-52.	3494.	965.	796.
	45.1	0.0	50.0	9350.	3933.	-52.	3654.	965.	796.
	0.176		49.9	10097.	3244.	-55.	3676.	1033.	855.
	90.	0.2219	50.0		67.27				
	681.1	0.9336	50.	1.1496	0.3717	[illegible]	[illegible]	0.0037	[illegible]

TABLE V.– Continued.

RUN 32 [illegible] TEST [illegible]

PT	[illegible] QPSF [illegible] TEMP [illegible]	[illegible] PSI [illegible]	[illegible] PED POS CHNG STR LA STR	LIFT [illegible]	DRAG DRAG [illegible] CD	SIDE SIDE [illegible] CY	PITCH PITCH [illegible] CM	YAW YAW [illegible] CN	ROLL ROLL [illegible] CROLL
3	117.6	-4.0	0.0	1010.	1125.	-23.	10106.	2099.	4425.
	44.9	0.0	49.9	1010.	1125.	-23.	10106.	2099.	4425.
	0.175		50.0	1059.	1174.	-24.	10593.	2255.	4640.
	79.	0.2265	50.0		25.28				
	674.2	0.9540	50.0	0.1288	0.1376	-0.0028	0.2359	0.0340	0.0164
4	117.6	-2.0	0.0	2505.	1074.	-54.	8704.	2036.	3491.
	44.9	0.0	49.9	2505.	1074.	-54.	8704.	2036.	3491.
	0.175		50.0	2751.	1150.	-62.	9323.	2155.	4055.
	79.	0.2264	49.9		24.13				
	674.2	0.9540	50.0	0.3209	0.1326	-0.0073	0.2040	0.0078	0.0149
5	118.2	0.0	0.0	4280.	1054.	-32.	6428.	2182.	3175.
	45.3	0.0	50.0	4280.	1054.	-32.	6428.	2182.	3175.
	0.176		50.0	4474.	1117.	-34.	6751.	2292.	3329.
	80.	0.2263	50.0		23.62				
	674.9	0.9522	50.0	0.5226	0.1375	-0.0039	0.1561	0.0653	0.0121
6	118.2	2.0	0.0	5960.	1081.	-65.	4056.	2220.	2753.
	45.0	0.0	50.0	5960.	1081.	-65.	4056.	2220.	2753.
	0.176		50.0	6259.	1135.	-71.	4456.	2332.	2891.
	80.	0.2263	50.1		24.02				
	674.9	0.9522	50.0	0.7311	0.1326	-0.0083	0.1047	0.0085	0.0105
7	118.3	4.0	0.0	7365.	1191.	-74.	2456.	2270.	2909.
	45.0	0.0	50.1	7365.	1191.	-74.	2456.	2270.	2909.
	0.176		50.0	7770.	1253.	-78.	2584.	2389.	3061.
	81.	0.2259	50.1		26.44				
	675.5	0.9504	50.0	0.9039	0.1461	-0.0091	0.0301	0.0487	0.0111
8	118.3	6.0	0.0	8975.	1300.	-30.	-162.	1909.	3357.
	45.0	0.0	50.0	8975.	1300.	-30.	-162.	1909.	3357.
	0.176		49.9	9443.	1368.	-42.	-171.	2008.	3532.
	81.	0.2259	50.0		28.84				
	675.5	0.9504	50.0	1.1027	0.1595	-0.0049	-0.0036	0.0675	0.0125
9	118.5	8.0	0.0	10590.	1469.	-42.	-2324.	1502.	3064.
	45.1	0.0	49.9	10590.	1469.	-42.	-2324.	1502.	3064.
	0.176		49.9	11452.	1544.	-44.	-2450.	1580.	3230.
	82.	0.2255	50.0		32.40				
	675.1	0.9487	50.0	1.2939	0.1801	-0.0051	-0.0592	0.0065	0.0117
10	118.2	10.0	0.0	11940.	1659.	-49.	-3715.	1368.	3859.
	44.9	0.0	50.0	11940.	1659.	-49.	-3715.	1368.	3859.
	0.176		50.0	12533.	1744.	-52.	-3910.	1463.	4047.
	82.	0.2255	50.0		36.45				
	676.1	0.9487	50.0	1.4830	0.2081	-0.0060	-0.0870	0.0055	0.0147
11	118.1	12.0	0.0	12595.	1788.	-67.	-3691.	1044.	2430.
	44.8	0.0	50.1	12595.	1788.	-67.	-3691.	1044.	2430.
	0.175		50.0	13223.	1885.	-71.	-3891.	1155.	2502.
	82.	0.2255	49.9		34.44				
	676.1	0.9487	50.0	1.5660	0.2207	-0.0083	-0.0867	0.0042	0.0093
12	118.3	12.0	0.0	13380.	1900.	-67.	-4905.	1230.	2060.
	44.9	0.0	50.1	13380.	1900.	-67.	-4905.	1230.	2060.
	0.176		50.0	14082.	2003.	-72.	-5171.	1297.	2115.
	82.	0.2255	50.0		42.27				
	676.1	0.9487	50.0	1.6422	0.2335	-0.0107	-0.1167	0.0047	0.0077
13	118.5	8.0	0.0	10175.	1495.	-43.	-1615.	1660.	15378.
	45.0	0.0	50.0	10175.	1495.	-43.	-1615.	1660.	15378.
	0.176		50.0	10769.	1574.	-44.	-1706.	1755.	16245.
	83.	0.2250	70.3		33.24				
	676.7	0.9466	50.0	1.2390	0.1837	-0.0014	-0.0378	0.0068	0.0587
14	118.4	6.0	0.0	9415.	1565.	-4.	-858.	1465.	24221.
	44.9	0.0	49.9	9415.	1565.	-4.	-858.	1465.	24221.
	0.176		50.0	10474.	1653.	-10.	-906.	1569.	25587.
	83.	0.2250	49.0		34.02				
	676.7	0.9466	50.1	1.2191	0.1928	-0.0011	-0.0201	0.0057	0.0925
15	118.2	-4.0	0.0	6905.	1708.	-91.	12192.	652.	445.
	44.7	0.0	49.9	6905.	1708.	-91.	12192.	652.	445.
	0.175		50.0	7294.	1804.	-96.	12876.	684.	523.
	83.	0.2250	50.0		38.15				
	676.7	0.9467	50.0	0.8527	0.2109	-0.0112	0.2865	0.0025	0.0019
16	118.4	-2.0	0.0	8440.	1750.	-58.	12140.	559.	807.
	44.8	0.0	50.0	8447.	1750.	-58.	12140.	559.	807.
	0.175		50.0	8832.	1852.	-61.	12844.	592.	854.
	84.	0.2248	50.5		39.03				
	677.4	0.9449	50.0	1.0400	0.2157	-0.0071	0.2847	0.0021	0.0031
17	118.5	0.0	0.0	9940.	1843.	-115.	10181.	846.	1150.
	44.9	0.0	50.1	9940.	1843.	-115.	10181.	846.	1150.
	0.176		49.9	10456.	1950.	-122.	10775.	895.	1263.
	84.	0.2246	50.5		41.00				
	677.4	0.9449	50.0	1.2199	0.2265	-0.0141	0.2341	0.0032	0.0043
18	118.5	2.0	0.0	11295.	1982.	-115.	7154.	875.	-181.
	44.9	0.0	50.0	11295.	1982.	-115.	7154.	875.	-181.
	0.176		49.9	11954.	2093.	-122.	7571.	926.	-192.
	84.	0.2246	50.5		44.10				
	677.4	0.9449	50.0	1.3866	0.2436	-0.0141	0.1625	0.0033	-0.0007
19	118.5	4.0	0.0	12715.	2137.	-167.	4594.	1403.	552.
	44.9	0.0	50.0	12715.	2137.	-167.	4594.	1403.	552.
	0.176		50.0	13457.	2262.	-177.	4862.	1485.	572.
	84.	0.2246	50.5		47.56				
	677.4	0.9449	50.0	1.5629	0.2627	-0.0192	0.1075	0.0056	[illegible]
20	118.5	6.0	0.0	14155.	2314.	-145.	1812.	1560.	-173.
	44.9	0.0	49.9	14155.	2314.	-145.	1812.	1560.	-173.
	0.176		50.0	14981.	2464.	-155.	1917.	1651.	-183.
	84.	0.2246	50.5		51.51				
	677.4	0.9449	50.0	1.7400	0.2846	-0.0178	0.0424	0.0190	-0.0020

TABLE V.– Continued.

RUN 32 TEST 525

PT	KNTS / QPSF / MACH / TEMP / VSNDRTS	ALPHA,C / PSI / HHUMIDTY / RHUM	INA / PED POS / LONG STK / LAT STK / POWLEV	LIFT,O / LIFT / LIFT/Q / CL	DRAG,O / DRAG / DRAG/Q / FE / CD	SIDE,O / SIDE / SIDE/Q / CY	PITCH,O / PITCH / PIT M/Q / CM	YAW,O / YAW / YAW/Q / CN	ROLL,O / ROLL / ROLL/Q / CROLL
21	118.4	4.0	0.0	15435.	2533.	-126.	-1496.	1152.	30.
	44.9	0.0	50.0	15435.	2533.	-126.	-1496.	1152.	30.
	0.176		49.9	16315.	2601.	-135.	-1595.	1219.	31.
	84.	0.2246	50.5		56.44				
	677.4	0.9949	50.0	1.9662	0.3114	-0.0158	-0.0351	0.0044	0.0001
22	118.4	10.0	0.0	16535.	2775.	-140.	-3339.	1062.	-453.
	44.9	0.0	49.5	16535.	2775.	-140.	-3339.	1062.	-453.
	0.176		50.0	17499.	2937.	-149.	-3534.	1145.	-1004.
	84.	0.2246	50.6		61.85				
	677.4	0.9949	50.0	2.0350	0.3416	-0.0172	-0.0782	0.0041	-0.0036
23	118.6	8.0	0.0	15280.	2467.	-211.	-519.	-132.	2915.
	44.9	0.0	49.9	15280.	2467.	-211.	-519.	-132.	2915.
	0.176		49.9	16201.	2615.	-224.	-550.	-140.	3088.
	85.	0.2242	70.1		54.84				
	678.0	0.9952	50.0	1.8782	0.3032	-0.0259	-0.0121	-0.0005	[illegible]
24	118.6	8.0	0.0	14500.	2366.	-270.	1485.	-1021.	7712.
	44.9	0.0	49.7	14500.	2366.	-270.	1485.	-1021.	7712.
	0.176		50.0	[illegible]	2509.	-293.	1574.	-1083.	8177.
	85.	0.2242	69.4		52.64				
	678.0	0.9952	50.0	1.7946	0.2405	-0.0334	0.0347	-0.0039	0.0295

RUN 33 TEST 525

PT	KNTS / QPSF / MACH / TEMP / VSNDRTS	ALPHA,C / PSI / HHUMIDTY / RHUM	INA / PED POS / LONG STK / LAT STK / POWLEV	LIFT,O / LIFT / LIFT/Q / CL	DRAG,O / DRAG / DRAG/Q / FE / CD	SIDE,O / SIDE / SIDE/Q / CY	PITCH,O / PITCH / PIT M/Q / CM	YAW,O / YAW / YAW/Q / CN	ROLL,O / ROLL / ROLL/Q / CROLL
4	116.1	-4.0	0.0	453.	1067.	-56.	10346.	2267.	5162.
	44.9	0.0	50.0	453.	1067.	-56.	10346.	2267.	5162.
	0.176		50.0	499.	1122.	-59.	10874.	2383.	5427.
	80.	0.2261	49.9		23.76				
	674.9	0.9513	50.0	[illegible]	0.1313	-0.0069	0.2423	0.0047	0.0197
5	117.9	-2.0	0.0	2725.	1014.	-58.	9455.	2162.	3810.
	44.8	0.0	49.9	2725.	1014.	-58.	9455.	2162.	3810.
	0.175		50.0	2984.	1066.	-61.	9519.	2273.	4005.
	80.	0.2261	50.0		22.64				
	674.9	0.9513	50.0	0.3382	0.1251	-0.0072	0.2170	0.0083	0.0146
6	118.0	0.0	0.0	4430.	992.	-74.	6739.	2191.	3435.
	44.8	0.0	50.0	4430.	992.	-70.	6739.	2191.	3435.
	0.175		50.0	4657.	1043.	-78.	7084.	2303.	3610.
	80.	0.2261	50.0		22.14				
	674.9	0.9513	50.0	0.5492	0.1229	-0.0091	0.1581	0.0084	0.0132
7	118.0	2.0	0.0	6105.	1049.	-95.	4241.	2443.	4120.
	44.9	0.0	50.0	6105.	1049.	-95.	4241.	2443.	4120.
	0.176		50.0	6416.	1093.	-100.	4458.	2568.	4339.
	80.	0.2261	50.0		23.17				
	674.9	0.9513	50.0	0.7510	0.1280	-0.0117	0.0993	0.0043	0.0156
8	118.2	4.0	0.0	7640.	1123.	-95.	2173.	2154.	3654.
	44.9	0.0	50.1	7640.	1123.	-95.	2173.	2154.	3654.
	0.176		50.0	8046.	1183.	-100.	2289.	2269.	3848.
	81.	0.2257	50.0		25.01				
	675.5	0.9493	50.0	[illegible]	0.1382	-0.0117	0.0509	0.0082	0.0140
9	118.3	6.0	0.0	9210.	1257.	-40.	-149.	1980.	4056.
	44.9	0.0	50.0	9210.	1257.	-40.	-149.	1980.	4056.
	0.175		50.0	9763.	1324.	-45.	-264.	2084.	4272.
	81.	0.2257	50.1		28.06				
	675.5	0.9496	50.0	1.1442	0.1551	-0.0111	-0.0047	0.0075	0.0156
10	118.2	8.0	0.0	10985.	1403.	-53.	-2682.	1392.	4160.
	44.9	0.0	49.9	10985.	1403.	-53.	-2682.	1392.	4160.
	0.175		50.0	11594.	1480.	-56.	-2830.	1469.	4390.
	82.	0.2253	50.1		31.24				
	676.1	0.9478	50.0	1.3537	0.1729	-0.0065	-0.0629	0.0053	0.0162
11	118.3	8.0	0.0	10585.	1431.	-102.	-2137.	1620.	16215.
	44.9	0.0	50.0	10585.	1431.	-102.	-2137.	1620.	16215.
	0.176		50.0	11164.	1513.	-108.	-2259.	1713.	17140.
	83.	0.2249	69.3		31.96				
	676.7	0.9460	49.9	1.3038	0.1763	-0.0126	-0.0501	0.0062	0.0621
12	118.3	8.0	0.0	10145.	1506.	-45.	-795.	1113.	27241.
	44.8	0.0	49.9	10145.	1506.	-45.	-795.	1113.	27241.
	0.175		50.0	10724.	1592.	-48.	-835.	1174.	28795.
	83.	0.2249	85.5		35.03				
	676.7	0.9460	50.0	1.2515	0.1858	-0.0056	-0.0185	0.0043	0.1045
13	118.1	8.0	0.0	15685.	2462.	-111.	-1172.	1444.	150.
	44.7	0.0	49.8	15685.	2462.	-111.	-1172.	1444.	150.
	0.175		50.0	16574.	2603.	-117.	-1244.	1526.	159.
	83.	0.2249	50.9		55.56				
	676.7	0.9461	50.0	1.9399	0.3070	-0.0137	-0.0277	0.0056	0.0006
14	119.0	8.0	0.0	14945.	2017.	-112.	338.	1054.	1452.
	45.2	0.0	49.6	14945.	2017.	-112.	338.	1054.	1452.
	0.176		49.4	15834.	2137.	-119.	359.	1116.	1538.
	84.	0.2243	50.7		44.58				
	677.3	0.9438	50.0	1.8208	0.2465	-0.0137	0.0079	0.0042	0.0055
15	119.0	8.0	0.0	14385.	1986.	-219.	1791.	706.	4396.
	45.2	0.0	49.7	14385.	1986.	-219.	1791.	706.	4396.
	0.176		49.4	15241.	2104.	-232.	1847.	750.	4655.
	84.	0.2243	70.2		43.4				
	677.4	0.9434	50.0	1.7575	0.2426	-0.0254	0.0416	0.0027	[illegible]
16	118.7	8.0	0.0	13220.	1925.	-247.	4011.	-340.	18030.
	45.0	0.0	49.8	13220.	1925.	-247.	4011.	-340.	18030.
	0.175		50.0	14005.	2039.	-262.	4249.	-360.	19117.
	84.	0.2244	[illegible]		42.76				
	677.4	0.9434	50.0	1.6233	0.2364	-0.0303	0.0957	-0.0013	0.0716

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OF POOR QUALITY

TABLE V.— Continued.

[illegible] 38 TEST 525

PT	[illegible]	ALPHA [illegible]	[illegible]	[illegible]	[illegible]	SIDE	PITCH	YAW	ROLL
3	78.0	-4.0	60.0	2445.	1626.	-1.	7544.	174.	174.
	20.3	0.0	50.0	2445.	1626.	-1.	7640.	174.	174.
	0.117		49.9	2444.	1650.	-1.	7545.	16[illegible].	176.
	65.	0.2345	50.0		62.05				
	655.4	0.9889	50.1	0.7444	0.0422	-0.0003	0.4052	0.0015	0.0015
4	78.0	-2.0	60.0	3445.	1672.	-4.	7176.	268.	634.
	20.3	0.0	50.0	3445.	1672.	-4.	7176.	268.	634.
	0.117		49.9	3449.	1694.	-4.	7212.	269.	643.
	65.	0.2345	49.9		82.24				
	655.0	0.9888	50.1	0.4368	0.4587	-0.0011	0.3715	0.0024	0.0054
5	78.0	0.0	60.0	3950.	1742.	-6.	5404.	61.	221.
	20.3	0.0	50.0	3950.	1742.	-6.	5404.	61.	221.
	0.117		49.8	4005.	1765.	-6.	5500.	61.	224.
	65.	0.2345	50.0		85.04				
	655.4	0.9888	49.9	1.0725	0.4731	-0.0016	0.3002	0.0005	0.0019
6	78.1	2.0	60.0	4465.	1812.	-2.	466.	103.	21.
	20.4	0.0	50.0	4465.	1812.	-2.	466.	103.	21.
	0.116		49.9	4525.	1834.	-2.	4725.	104.	21.
	65.	0.2345	49.9		85.47				
	655.0	0.9888	50.1	1.2044	0.4910	-0.0005	0.2473	0.0004	0.0032
7	75.0	4.0	60.0	4920.	1912.	-13.	3479.	128.	-191.
	20.3	0.0	49.9	4920.	1912.	-13.	3479.	128.	-191.
	0.117		49.9	4960.	1930.	-13.	3525.	130.	-194.
	65.	0.2345	50.0		94.11				
	665.4	0.9888	50.1	1.3374	0.5199	-0.0035	0.1900	0.0011	-0.0015
8	78.3	6.0	60.0	5445.	2014.	-15.	2183.	304.	-172.
	20.4	0.0	50.0	5445.	2014.	-15.	2183.	304.	-172.
	0.116		50.1	5540.	2044.	-15.	2221.	345.	-175.
	67.	0.2337	49.8		94.78				
	666.7	0.9830	50.0	1.4944	0.5457	-0.0041	0.1120	0.0031	-0.0014
9	78.2	8.0	60.0	5935.	2124.	-24.	730.	471.	-91.
	20.3	0.0	50.0	5935.	2124.	-23.	730.	471.	-91.
	0.117		50.1	6037.	2161.	-24.	751.	479.	-92.
	67.	0.2337	50.0		104.42				
	666.7	0.9830	50.0	1.6120	0.5769	-0.0065	0.0382	0.0040	-0.0008
10	78.2	10.0	60.0	6345.	2222.	-34.	191.	431.	75.
	20.3	0.0	50.0	6345.	2222.	-39.	191.	431.	75.
	0.117		50.0	6455.	2260.	-40.	195.	438.	76.
	67.	0.2337	50.0		107.21				
	666.7	0.9830	50.0	1.7233	0.6035	-0.0106	0.0099	0.0036	0.0006
11	78.3	12.0	60.0	6545.	2268.	1.	-59.	217.	-480.
	20.4	0.0	50.0	6545.	2268.	1.	-59.	217.	-480.
	0.115		50.1	6671.	2312.	1.	-60.	221.	-488.
	68.	0.2332	50.0		111.3[illegible]				
	667.3	0.9812	49.9	1.7756	0.6153	0.0003	-0.0020	0.0018	-0.0039
12	99.3	-4.0	60.0	4640.	2623.	-12.	12015.	482.	467.
	32.6	0.0	49.4	4640.	2623.	-12.	12015.	482.	467.
	0.149		50.0	4800.	2685.	-12.	12267.	484.	475.
	65.	0.2322	50.0		80.37				
	667.3	0.9770	50.0	0.7939	0.4460	-0.0020	0.3949	-0.0004	0.0025
13	99.6	-2.0	60.0	5515.	2703.	-10.	11744.	171.	-60.
	32.8	0.0	49.9	5515.	2703.	-10.	11744.	171.	-60.
	0.150		50.1	5666.	2777.	-10.	12070.	176.	-61.
	70.	0.2313	50.0		82.44				
	668.6	0.9733	50.0	0.9289	0.4553	-0.0017	0.3745	0.0009	-0.0003
14	99.6	0.0	60.0	6305.	2811.	-3.	9841.	106.	-134.
	32.7	0.0	49.9	6305.	2811.	-3.	9841.	106.	-134.
	0.149		50.1	6490.	2894.	-3.	10134.	109.	-138.
	71.	0.2309	49.9		85.69				
	669.2	0.9715	50.0	1.0682	0.4745	-0.0005	0.316[illegible]	0.0006	-0.000[illegible]
15	99.9	2.0	60.0	7180.	2933.	-23.	8488.	505.	-89.
	32.7	0.0	50.0	7180.	2933.	-23.	8488.	505.	-89.
	0.149		50.0	7405.	3025.	-24.	8755.	521.	-91.
	72.	0.2305	50.0		89.61				
	669.4	0.9697	49.9	1.2119	0.4951	-0.0039	0.2726	0.0027	-0.0005
16	100.1	4.0	60.0	8095.	3099.	-20.	5180.	569.	-894.
	32.8	0.0	49.9	8095.	3099.	-20.	5180.	569.	-894.
	0.150		50.2	8364.	3197.	-2[illegible].	5365.	586.	-927.
	73.	0.2300	50.1		9[illegible].1[illegible]				
	670.5	0.9678	50.0	1.3511	0.52[illegible]	-0.0034	0.1972	0.0030	-0.0047
17	99.9	6.0	60.0	8895.	3239.	-31.	3695.	740.	-541.
	32.7	0.0	49.9	8895.	3239.	-31.	3695.	740.	-541.
	0.149		50.1	9191.	3347.	-32.	3818.	816.	-552.
	73.	0.2300	50.0		99.03				
	670.5	0.9678	49.9	1.5025	0.5471	-0.0052	0.1185	0.0041	-0.0012
18	99.8	8.0	60.0	9250.	3355.	-10.	2801.	519.	-1018.
	32.7	0.0	50.0	9250.	3355.	-10.	2801.	519.	-1018.
	0.149		50.1	9597.	3480.	-10.	2894.	535.	-1051.
	73.	0.2301	50.0		102.11				
	670.5	0.9679	49.9	1.5597	0.5652	-0.00[illegible]	0.0902	0.0027	-0.0054
19	99.9	8.0	[illegible]	9675.	3460.	-2.	1591.	822.	-354.
	32.8	0.0	50.0	9675.	3460.	-2.	1591.	822.	-357.
	0.149		50.1	100[illegible]5.	3527.	-2.	1[illegible].	851.	-370.
	74.	0.2296	49.9		104.39				
	671.1	0.9682	50.0	1.6375	0.5787	-0.00[illegible]	[illegible]	0.0044	-0.00[illegible]
20	100.3	9.0	[illegible]	10495.	3525.	1.	[illegible]	26[illegible].	-1374.
	32.9	0.0	50.0	10495.	3525.	1.	[illegible]	267.	-1374.
	0.150		[illegible]	10[illegible].	3[illegible].	1.	[illegible]	270.	-1425.
	74.	0.2295	[illegible]		[illegible]				
	671.1	0.9660	49.7	1.6992	[illegible]	0.0017	[illegible]	0.0018	-0.00[illegible]

TABLE V.– Continued.

RUN 34 TEST 525

PT	VATS QPSF MTUN TEMP VSNDFTS	ALPHA,C PSI RHOx100 RHOR	INA PED POS LONG STK LAT STK PCT DEV	LIFT,U LIFT LIFT/E CL	DRAG,U DRAG DRAG/E FE CD	SIDE,U SIDE SIDE/E CY	PITCH,U PITCH PITCH/E CM	YAW,U YAW YAW/E CN	ROLL,U ROLL ROLL/E CROLL
21	115.7	-4.0	60.0	6355.	3576.	-26.	17395.	131.	164.
	43.5	0.0	50.0	6355.	3576.	-26.	17395.	131.	164.
	0.174		50.2	6617.	3724.	-27.	18105.	135.	170.
	75.	0.2283	50.0		60.21				
	673.7	0.9600	49.9	0.7958	0.4464	-0.0032	0.4150	0.0015	0.0027
22	116.5	-2.0	60.0	7910.	3661.	-42.	16652.	206.	-322.
	44.0	0.0	50.0	7910.	3661.	-42.	16652.	206.	-322.
	0.174		50.0	7750.	3819.	-44.	17344.	215.	-336.
	76.	0.2279	50.0		63.18				
	672.9	0.9587	50.0	0.9300	0.4595	-0.0053	0.3534	0.0026	-0.0013
23	116.4	0.0	60.0	8570.	3775.	-12.	13879.	130.	25.
	44.3	0.0	50.0	8570.	3775.	-12.	13879.	130.	25.
	0.174		50.0	8939.	3936.	-13.	14472.	136.	26.
	76.	0.2279	50.0		65.04				
	672.9	0.9587	49.9	1.0767	0.4703	-0.0040	0.3316	0.0015	0.0001
24	116.3	2.0	60.0	9592.	3950.	-39.	11352.	370.	-336.
	43.8	0.0	50.0	9592.	3950.	-39.	11352.	370.	-336.
	0.173		50.2	10074.	4126.	-41.	11963.	385.	-351.
	77.	0.2275	50.0		90.15				
	673.0	0.9569	50.0	1.2156	0.4981	-0.0049	0.2724	0.0013	-0.0013
25	116.0	4.0	60.0	10690.	4161.	-44.	7669.	641.	177.
	43.6	0.0	50.0	10690.	4161.	-44.	7669.	641.	177.
	0.173		50.3	11170.	4327.	-46.	8222.	670.	185.
	77.	0.2275	50.0		94.95				
	673.0	0.9570	[illegible]	1.3594	0.5207	-0.0056	0.1547	0.0025	0.0007
26	115.6	6.0	60.0	11880.	4335.	-31.	5635.	571.	-139.
	43.6	0.0	50.0	11880.	4335.	-31.	5635.	571.	-139.
	0.173		50.2	12394.	4538.	-33.	5880.	597.	-145.
	78.	0.2271	50.0		99.89				
	673.6	0.9553	49.9	1.5073	0.5519	-0.0042	0.1218	0.0023	-0.0005
27	115.7	7.0	60.0	12310.	4461.	-57.	3748.	729.	-1286.
	43.3	0.0	50.0	12310.	4461.	-57.	3748.	729.	-1286.
	0.172		50.0	12866.	4670.	-60.	3921.	762.	-1346.
	78.	0.2271	50.0		103.01				
	673.6	0.9553	49.9	1.6705	0.5891	-0.0075	0.0914	0.0028	-0.0051
28	115.4	8.0	60.0	12790.	4547.	-57.	2315.	856.	-570.
	43.3	0.0	50.1	12790.	4547.	-57.	2315.	856.	-570.
	0.172		50.1	13334.	4759.	-60.	2423.	896.	-596.
	78.	0.2271	50.0		105.02				
	673.6	0.9554	49.9	1.6350	0.5835	-0.0073	0.0565	0.0034	-0.0023

RUN 35 TEST 525

PT	VATS QPSF MTUN TEMP VSNDFTS	ALPHA,C PSI RHOx100 RHOR	INA PED POS LONG STK LAT STK PCT DEV	LIFT,U LIFT LIFT/E CL	DRAG,U DRAG DRAG/E FE CD	SIDE,U SIDE SIDE/E CY	PITCH,U PITCH PITCH/E CM	YAW,U YAW YAW/E CN	ROLL,U ROLL ROLL/E CROLL
3	116.9	-4.0	60.0	6710.	3538.	-32.	17165.	541.	1390.
	43.8	0.0	49.9	6710.	3538.	-32.	17165.	541.	1390.
	0.176		50.0	7087.	3737.	-34.	18150.	572.	1466.
	86.	0.2251	49.9		60.71				
	676.1	0.9468	49.9	0.8457	0.4459	-0.0040	0.4122	0.0021	0.0054
4	116.8	-2.0	60.0	7800.	3654.	-36.	16210.	769.	1072.
	43.7	0.0	50.0	7800.	3654.	-36.	16210.	769.	1072.
	0.175		50.0	8238.	3859.	-38.	17119.	812.	1132.
	86.	0.2251	50.0		63.62				
	676.1	0.9469	50.0	0.9882	0.4620	-0.0046	0.3900	0.0030	0.0042
5	116.6	0.0	60.0	8775.	3789.	-12.	14422.	727.	986.
	43.6	0.0	49.9	8775.	3789.	-12.	14422.	727.	986.
	0.173		50.0	9267.	4001.	-13.	15231.	767.	1041.
	86.	0.2251	50.0		66.48				
	676.1	0.9469	49.9	1.1131	0.4806	-0.0015	0.3481	0.0028	0.0039
6	116.0	4.0	60.0	11090.	4181.	-52.	8306.	923.	695.
	43.3	0.0	50.0	11090.	4181.	-52.	8306.	923.	695.
	0.173		50.0	11701.	4384.	-55.	8765.	974.	736.
	86.	0.2243	50.0		95.62				
	677.4	0.9435	49.9	1.4085	0.5283	-0.0066	0.2016	0.0037	0.0028
7	116.3	6.0	60.0	12055.	4359.	-30.	5041.	852.	621.
	43.1	0.0	49.9	12055.	4359.	-30.	5041.	852.	621.
	0.172		50.0	12794.	4626.	-32.	5352.	904.	659.
	85.	0.2239	50.0		101.15				
	678.0	0.9419	49.9	1.5450	0.5588	-0.0038	0.1230	0.0034	0.0025
8	117.4	-6.0	60.0	1865.	2907.	8.	15811.	1439.	3368.
	44.3	0.0	49.8	1865.	2907.	8.	15811.	1439.	3368.
	0.175		50.0	1981.	3088.	8.	16795.	1529.	3578.
	85.	0.2230	50.1		65.59				
	678.0	0.9414	50.0	0.2325	0.3624	0.0037	0.3750	0.0056	0.0131
9	117.8	[illegible]	60.0	4370.	2998.	15.	11406.	3577.	3012.
	44.3	0.0	49.8	4370.	2998.	15.	11406.	3577.	3012.
	0.174		50.0	4642.	3182.	16.	12115.	[illegible]	3200.
	85.	0.2234	50.0		67.70				
	678.0	0.9415	49.9	0.5450	0.3740	0.0019	0.2710	0.0061	0.0117
10	116.1	4.0	60.0	6485.	3133.	-35.	6993.	1063.	1606.
	46.3	0.0	50.0	6485.	3133.	-35.	6993.	1063.	1606.
	0.174		50.0	6913.	3340.	-37.	7455.	1134.	1713.
	87.	0.2230	49.9		70.56				
	679.2	0.9380	50.0	0.8096	0.3911	-0.0044	0.1661	0.0041	0.0062
11	117.9	8.0	60.0	7625.	3290.	30.	4542.	890.	2355.
	44.1	0.0	49.9	7625.	3290.	30.	4542.	890.	2355.
	0.174		50.0	7915.	3507.	32.	4840.	949.	2510.
	87.	0.2234	49.9		74.62				
	679.2	0.9381	50.0	0.9304	0.4122	0.0038	0.1107	0.0035	0.0092

TABLE V.— Continued.

RUN 36 TEST 525

PT	VKTS DPSF MTON TEMP VSNDKTS	ALPS,L PSI MACH,IDL RHOW	1HA PED POS LONG STK LAT STK POW LEV	LIFT,O LIFT LIFT/Z CL	DRAG,O DRAG DRAG/D FE CD	SIDE,O SIDE SIDE/Q CY	PITCH,O PITCH PITCH/Q CM	YAW,O YAW YAW/Q CN	ROLL,O ROLL ROLL/Q CROLL
3	116.7	[illegible]	60.[illegible]	6260.	3031.	-82.	16823.	1059.	465.
	84.8	0.0	50.[illegible]	6260.	3031.	-82.	16823.	1059.	465.
	0.176		50.2	6444.	3119.	-84.	17306.	1099.	565.
	65.	0.2316	49.4		67.6[illegible]				
	867.3	0.9719	49.9	0.7715	0.3735	-0.0101	0.3445	0.0041	0.0035
4	116.9	0.0	60.0	8540.	3294.	-64.	14784.	47.	-540.
	84.5	0.0	50.3	8540.	3294.	-64.	14784.	47.	-540.
	0.176		50.1	8820.	3402.	-66.	15204.	49.	-463.
	70.	0.2331	49.4		[illegible]				
	865.6	0.9683	50.1	1.0550	0.4067	-0.0029	0.3466	0.0012	-0.0015
5	117.2	2.0	60.	9760.	3468.	-3.	12021.	1204.	386.
	84.6	0.0	50.1	9760.	3468.	-3.	12021.	1208.	386.
	0.176		50.1	10074.	3569.	-3.	12426.	1250.	402.
	71.	0.2292	50.0		77.40				
	869.2	0.9654	50.0	1.2009	0.4276	-0.0004	0.2824	0.0046	0.0015
6	117.2	4.0	60.0	10655.	3679.	-41.	7746.	1602.	854.
	85.0	0.0	50.1	10655.	3679.	-41.	7746.	1602.	854.
	0.176		50.1	11026.	3807.	-42.	8017.	1657.	884.
	71.	0.2297	49.9		81.61				
	869.2	0.9654	50.1	1.3090	0.4520	-0.0050	0.1841	0.0061	0.0033
7	117.2	4.0	60.0	1355.	2435.	-6.	15444.	1771.	2566.
	84.4	0.0	50.1	1355.	2435.	-6.	15444.	1771.	2566.
	0.176		50.1	1467.	2524.	-5.	16045.	1844.	2503.
	75.	0.2284	50.1		54.34				
	870.5	0.9628	50.1	0.1871	0.3002	-0.0007	0.3675	0.0066	0.0094
8	117.3	6.0	60.	3870.	2517.	27.	12161.	1798.	2427.
	84.8	0.0	50.2	3870.	2517.	27.	12161.	1798.	2427.
	0.176		50.1	4014.	2614.	25.	12631.	1867.	2521.
	75.	0.2284	50.0		56.14				
	870.5	0.9628	49.9	0.4789	0.3102	0.0035	0.2852	0.0064	0.0093
9	117.6	2.0	60.0	5080.	2636.	15.	10251.	1690.	2246.
	85.0	0.0	50.0	5080.	2636.	15.	10251.	1690.	2246.
	0.176		50.1	5266.	2745.	16.	10666.	1765.	2337.
	74.	0.2284	50.0		58.65				
	871.1	0.9610	50.0	0.6241	0.3240	0.0018	0.2397	0.0065	0.0086
10	117.6	4.0	60.0	6155.	2780.	62.	8401.	1223.	2145.
	84.9	0.0	50.2	6155.	2780.	62.	8401.	1223.	2145.
	0.176		50.1	6417.	2890.	65.	8457.	1275.	2237.
	75.	0.2280	49.4		61.48				
	871.7	0.9592	49.4	0.7570	0.3419	0.0076	0.2013	0.0047	0.0082
11	117.3	5.0	60.0	6780.	2830.	55.	7515.	1722.	2401.
	85.7	0.0	50.1	6780.	2830.	55.	7515.	1722.	2401.
	0.175		50.1	7020.	2950.	57.	7834.	1802.	2424.
	75.	0.2280	50.0		63.32				
	871.7	0.9593	50.0	0.8332	0.3498	0.0068	0.1760	0.0066	0.0104
12	117.4	6.0	60.0	7145.	2898.	41.	7028.	1667.	2700.
	84.8	0.0	50.0	7145.	2898.	41.	7028.	1667.	2700.
	0.175		50.0	7457.	3023.	43.	7327.	1737.	2795.
	75.	0.2240	50.0		64.71				
	871.7	0.9592	49.9	0.9961	0.3575	0.0051	0.1650	0.0064	0.0116

RUN 37 TEST 525

PT	VKTS DPSF MTON TEMP VSNDKTS	ALPS,L PSI MACH,IDL RHOW	1HA PED POS LONG STK LAT STK POW LEV	LIFT,O LIFT LIFT/Z CL	DRAG,O DRAG DRAG/D FE CD	SIDE,O SIDE SIDE/Q CY	PITCH,O PITCH PITCH/Q CM	YAW,O YAW YAW/Q CN	ROLL,O ROLL ROLL/Q CROLL
3	118.1	4.0	60.0	6070.	3041.	32.	15674.	363.	455.
	84.8	0.0	50.0	6070.	3041.	32.	15978.	363.	455.
	0.175		50.0	6401.	3207.	34.	16744.	383.	480.
	81.	0.2254	50.0		67.46				
	875.5	0.9483	49.9	[illegible]	0.3751	0.0039	0.3727	0.0014	0.0032
4	117.8	0.0	60.0	8385.	3276.	24.	14093.	664.	703.
	84.6	0.0	50.1	8385.	3276.	24.	14093.	664.	700.
	0.175		50.0	8842.	3454.	25.	14861.	700.	739.
	81.	0.2254	49.4		73.53				
	875.5	0.9483	49.9	[illegible]	0.4052	0.0030	0.3325	0.0026	0.0027
5	117.6	2.0	60.0	9520.	3434.	42.	11917.	994.	-151.
	84.5	0.0	50.0	9520.	3434.	42.	11917.	994.	-151.
	0.175		50.0	10057.	3625.	44.	12564.	1050.	-160.
	82.	0.2250	53.0		77.19				
	876.1	0.9466	49.9	1.1823	0.4265	0.0052	0.2815	0.0036	-0.0006
6	117.6	4.0	60.0	10580.	3641.	-13.	9366.	1103.	-437.
	84.2	0.0	49.9	10580.	3641.	-13.	9366.	1103.	-437.
	0.174		51.	11175.	3855.	-14.	[illegible]	1233.	-462.
	83.	0.2246	50.		82.36				
	876.7	0.9450	49.9	1.3197	0.4557	-0.0016	0.2227	0.0045	-0.0017
7	99.5	4.0	75.0	3905.	2342.	7.	11051.	-447.	-522.
	51.8	0.0	49.9	3905.	2342.	7.	11051.	-447.	-522.
	0.147		50.	4115.	2510.	7.	11645.	-449.	-551.
	83.	0.2256	50.1		74.05				
	876.7	0.9450	49.4	0.8777	0.4134	0.0012	0.3658	-0.0015	-0.0028
8	44.6	0.0	75.0	5550.	2493.	5.	10034.	432.	-413.
	11.9	0.0	50.2	5555.	2493.	5.	10036.	432.	-414.
	0.148		50.0	5848.	2627.	5.	10575.	455.	-462.
	83.	0.2255	49.4		78.26				
	876.7	0.9440	49.9	0.9649	0.8325	0.0004	0.3313	0.0025	-0.0044

TABLE V.— Continued.

RUN 37 TEST 525

PT	VKTS QPSF RUN TEMP VSNDKTS	ALFS,E PSI RHO,SLUG RHOR	INA PED POS LONG STA LAT STA PDW CEY	LIFT,LB LIFT LIFT/6 CL	DRAG,LB DRAG DRAG/6 FE CD	SIDE,LB SIDE SIDE/6 CY	PITCH,LB PITCH PITCH/6 CM	YAW,LB YAW YAW/6 CN	ROLL,LB ROLL ROLL/6 CROLL
9	49.5	2.0	75.0	6555.	2598.	-7.	8602.	80.	-740.
	31.8	0.0	49.9	6555.	2598.	-7.	8462.	80.	-740.
	0.147		50.0	6675.	2738.	-7.	9538.	91.	-780.
	83.	0.2256	50.0		61.73				
	676.7	0.9490	49.8	1.1011	0.4516	-0.0012	0.2951	0.0005	-0.0040
10	49.5	4.0	75.0	7170.	2726.	-25.	7193.	220.	-593.
	31.8	0.0	49.9	7170.	2726.	-25.	7193.	220.	-593.
	0.147		50.0	7555.	2875.	-27.	7579.	232.	-625.
	85.	0.2256	50.0		65.76				
	676.7	0.9490	49.8	1.2453	0.4738	-0.0045	0.2377	0.0012	-0.0032
11	156.6	-4.0	0.0	1455.	2209.	-46.	18230.	4003.	7716.
	76.9	0.0	49.9	1455.	2209.	-46.	18230.	4003.	7716.
	0.232		50.1	1569.	2382.	-50.	19661.	4317.	8321.
	87.	0.2204	49.9		26.71				
	679.2	0.9273	49.9	0.1035	0.1586	-0.0033	0.2494	0.0064	0.0172
12	156.5	0.0	0.0	6905.	2132.	-30.	12173.	3766.	7654.
	76.8	0.0	50.0	6905.	2132.	-30.	12173.	3766.	7654.
	0.232		50.0	7460.	2303.	-37.	13152.	4069.	8455.
	88.	0.2200	50.1		27.76				
	679.9	0.9256	49.9	0.4968	0.1534	-0.0024	0.1567	0.0084	0.0176
13	156.9	4.0	0.0	12380.	2338.	-55.	5037.	2934.	7713.
	76.9	0.0	50.0	12380.	2338.	-55.	5037.	2934.	7713.
	0.232		50.0	13426.	2535.	-63.	5424.	3182.	8364.
	90.	0.2192	50.1		30.41				
	681.1	0.9222	49.8	0.8897	0.1680	-0.0042	0.0685	0.0066	0.0172
14	157.0	6.0	0.0	15275.	2590.	-85.	714.	2197.	7425.
	76.8	0.0	50.0	15275.	2590.	-85.	714.	2197.	7425.
	0.232		50.1	16576.	2810.	-92.	[illegible]	[illegible]	8064.
	91.	0.2158	50.0		33.77				
	681.7	0.9226	49.8	1.0843	0.1866	-0.0061	0.0197	0.0065	0.0166
15	157.2	8.0	0.0	17720.	2924.	-121.	-3614.	2087.	5652.
	77.0	0.0	50.0	17720.	2924.	-121.	-3614.	2087.	5652.
	0.232		50.0	19250.	3176.	-131.	-3926.	2267.	6140.
	91.	0.2166	49.9		37.94				
	681.7	0.9205	49.8	1.2721	0.2099	-0.0087	-0.0494	0.0047	0.0126
16	157.0	10.0	0.0	20030.	3277.	-113.	-6276.	1274.	5755.
	76.7	0.0	49.9	20030.	3277.	-113.	-6274.	1274.	5755.
	0.231		50.0	21790.	3565.	-123.	-6830.	1386.	6261.
	92.	0.2155	49.9		42.71				
	682.3	0.9192	49.7	1.4425	0.2360	-0.0081	-0.0460	0.0029	0.0124

RUN 39 TEST 525

PT	VKTS QPSF RUN TEMP VSNDKTS	ALFS,E PSI RHO,SLUG RHOR	INA PED POS LONG STA LAT STA PDW CEY	LIFT,LB LIFT LIFT/6 CL	DRAG,LB DRAG DRAG/6 FE CD	SIDE,LB SIDE SIDE/6 CY	PITCH,LB PITCH PITCH/6 CM	YAW,LB YAW YAW/6 CN	ROLL,LB ROLL ROLL/5 CROLL
4	114.0	-4.0	60.0	5725.	3033.	-136.	16479.	776.	-222.
	43.4	0.0	50.0	5725.	3033.	-136.	16479.	776.	-222.
	0.172		50.0	5864.	3107.	-139.	16880.	794.	-228.
	67.	0.2322	50.2		64.49				
	668.7	0.9752	49.9	0.7289	0.3861	-0.0173	0.3492	0.0031	-0.0009
5	114.0	0.0	60.0	7925.	3261.	-122.	14339.	309.	-561.
	43.3	0.0	50.0	7925.	3261.	-122.	14339.	309.	-561.
	0.172		50.0	8144.	3353.	-125.	14743.	317.	-577.
	69.	0.2312	50.3		75.36				
	668.0	0.9728	49.9	1.0122	0.4165	-0.0156	0.3465	0.0012	-0.0025
6	49.3	-4.0	75.0	3675.	2467.	-66.	11217.	404.	-367.
	31.9	0.0	50.0	3675.	2467.	-66.	11217.	404.	-367.
	0.147		50.1	3771.	2476.	-56.	11511.	415.	-377.
	70.	0.2316	50.2		75.52				
	658.6	0.9705	49.9	0.6369	0.4171	-0.0149	0.3719	0.0022	-0.0020
7	49.2	0.0	75.0	5375.	2520.	-90.	10161.	389.	-856.
	31.8	0.0	50.1	5375.	2520.	-90.	10161.	389.	-856.
	0.147		50.0	5525.	2591.	-93.	10445.	400.	-879.
	71.	0.2312	50.1		74.28				
	669.2	0.9728	49.9	0.9362	0.4389	-0.0156	0.3161	0.0021	-0.0046
8	48.2	4.0	75.0	6980.	2697.	-10.	7553.	685.	-112.
	31.7	0.0	50.2	6980.	2697.	-10.	7553.	685.	-112.
	0.147		50.0	7176.	2773.	-10.	7766.	702.	-115.
	71.	0.2311	50.2		65.04				
	649.2	0.9725	50.0	1.2159	0.4698	-0.0017	0.2564	0.0037	-0.0006
9	116.7	-4.0	60.0	2105.	2469.	-22.	15213.	1334.	859.
	46.5	0.0	50.0	2105.	2469.	-22.	15213.	1334.	859.
	0.175		50.1	2146.	2560.	-23.	15772.	1383.	891.
	73.	0.2293	50.3		55.98				
	670.5	0.9646	50.0	0.1173	0.4093	-0.0027	0.3557	0.0032	0.0033
10	116.5	0.0	60.0	3525.	2535.	-56.	11410.	1305.	1981.
	46.3	0.0	50.2	3525.	2535.	-56.	11410.	1305.	1981.
	0.174		50.1	3654.	2628.	-58.	11828.	1353.	2053.
	73.	0.2243	50.2		57.22				
	670.5	0.9646	50.0	0.4356	0.3161	-0.0070	0.2767	0.0051	0.0078
11	116.5	2.0	60.0	4650.	2624.	-47.	9224.	1030.	1744.
	46.1	0.0	50.0	4650.	2624.	-47.	9224.	1030.	1744.
	0.174		50.1	4830.	2730.	-49.	9613.	1072.	1815.
	75.	0.2244	50.3		59.45				
	671.7	0.9643	50.0	0.5820	0.3284	-0.0059	0.2322	0.0040	0.0066

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TABLE V.— Continued.

RUN 39 TEST 525

PT	VATS QPSF MION TEMP VSNORTS	ALPS,U PSI ALPHA,U RHOR	JNA PED POS CONS STA LAT STA RUN LEV	LIFT,U LIFT LIFT/Q CL	DRAG,U DRAG DRAG/Q FE CD	SIDE,U SIDE SIDE/Q CY	PITCH,U PITCH PITCH/Q CM	YAW,U YAW YAW/Q CN	ROLL,U ROLL ROLL/Q CROLL
12	116.5	0.0	60.0	5805.	2742.	-33.	8269.	1386.	1767.
	88.0	0.0	50.2	5805.	2742.	-33.	8269.	1386.	1767.
	0.174		50.0	6903.	2858.	-34.	8822.	1445.	1842.
	78.	0.2281	50.3		62.25				
	672.6	0.9593	50.0	0.7281	0.3439	-0.0041	0.1974	0.0354	[illegible]
13	115.9	0.0	[illegible]	8965.	1963.	-152.	10300.	1195.	1061.
	85.5	0.0	51.2	8965.	1963.	-152.	10300.	1195.	1061.
	0.173		50.0	9375.	2033.	-159.	10772.	1250.	1109.
	76.	0.2273	50.3		45.10				
	673.6	0.9567	50.0	1.1389	0.2494	-0.0193	0.2490	0.0047	0.0042
14	115.8	0.0	50.0	8790.	2090.	-184.	10726.	1044.	544.
	83.4	0.0	50.0	8790.	2090.	-184.	10726.	1044.	544.
	0.172		50.1	9204.	2609.	-193.	11237.	1094.	622.
	79.	0.2259	50.4		57.44				
	672.2	0.9535	50.1	1.1282	0.3173	-0.0234	0.2603	0.0041	0.0024
15	115.9	0.0	50.0	8570.	2933.	-171.	12367.	-417.	-211.
	83.4	0.0	50.0	8570.	2933.	-171.	12367.	-417.	-211.
	0.173		50.1	8974.	3073.	-179.	12957.	-437.	-221.
	79.	0.2264	50.4		67.55				
	672.2	0.9545	50.1	1.0988	0.3732	-0.0218	0.2994	-0.0017	-0.0008
16	115.9	0.0	60.0	[illegible]	3209.	-136.	13707.	-71.	-1247.
	43.3	0.0	50.2	[illegible]	3209.	-136.	13707.	-71.	-1247.
	0.172		50.2	8470.	3366.	-143.	14367.	-75.	-1309.
	80.	0.2265	50.4		74.66				
	676.9	0.9527	50.1	1.0290	0.4092	-0.0173	0.3326	-0.0005	-0.0049
17	116.2	0.0	90.0	6755.	3395.	-163.	14174.	637.	182.
	43.3	0.0	50.1	6755.	3395.	-163.	14174.	637.	182.
	0.173		50.1	7090.	3553.	-166.	14878.	670.	191.
	80.	0.2265	50.3		77.46				
	674.9	0.9527	50.2	0.8595	0.4317	-0.0131	0.3432	0.0033	[illegible]

RUN 40 TEST 525

PT	VATS QPSF MION TEMP VSNORTS	ALPS,U PSI ALPHA,U RHOR	JNA PED POS CONS STA LAT STA RUN LEV	LIFT,U LIFT LIFT/Q CL	DRAG,U DRAG DRAG/Q FE CD	SIDE,U SIDE SIDE/Q CY	PITCH,U PITCH PITCH/Q CM	YAW,U YAW YAW/Q CN	ROLL,U ROLL ROLL/Q CROLL
3	117.2	[illegible]	60.0	5870.	2972.	-103.	15620.	193.	-595.
	44.1	0.0	50.0	5870.	2972.	-103.	15620.	193.	-595.
	0.174		50.0	6194.	3134.	-109.	16496.	203.	-625.
	83.	0.2251	[illegible]		67.44				
	676.7	[illegible]	49.9	0.7339	0.3726	-0.0129	0.3726	0.0008	-0.0023
4	117.5	0.0	60.0	8295.	3182.	-80.	13566.	467.	-520.
	44.1	0.0	50.0	8295.	3182.	-80.	13566.	467.	-520.
	0.174		50.1	8739.	3371.	-83.	14174.	480.	-551.
	85.	0.2242	50.1		72.20				
	678.0	0.9436	49.9	1.0337	0.3989	-0.0110	0.3236	[illegible]	-0.0026
5	117.6	[illegible]	60.0	9055.	3368.	-65.	11565.	614.	-452.
	44.1	0.0	50.0	9055.	3368.	-65.	11565.	614.	-452.
	0.174		50.1	10040.	3576.	-69.	[illegible]	652.	[illegible]
	86.	0.2236	50.0		75.39				
	678.6	0.9417	49.9	[illegible]	0.4220	-0.0051	[illegible]	0.0024	-0.0019
6	117.6	0.0	60.0	10595.	3381.	-93.	8085.	660.	259.
	44.0	0.0	49.8	10595.	3381.	-93.	8085.	660.	259.
	0.174		50.1	11204.	3704.	-88.	[illegible]	702.	276.
	87.	0.2234	[illegible]		60.96				
	679.2	0.9397	49.8	1.3298	0.4470	-0.0104	0.2276	0.0026	[illegible]
7	118.0	[illegible]	60.0	10207.	2392.	21.	14704.	650.	2236.
	44.1	0.0	49.8	10207.	2392.	21.	14704.	650.	2236.
	0.174		50.1	[illegible]	2555.	22.	15706.	694.	[illegible]
	[illegible]	0.2225	49.9		54.22				
	680.5	0.9362	49.9	[illegible]	0.2996	0.0026	0.3504	0.0025	[illegible]
8	118.0	0.0	60.0	3515.	2458.	-62.	11244.	888.	1675.
	44.1	0.0	49.9	3515.	[illegible]	-62.	11244.	888.	1675.
	0.174		50.0	3754.	2625.	-66.	12014.	944.	2003.
	89.	0.2225	50.0		55.69				
	680.5	0.9362	49.8	[illegible]	0.3077	[illegible]	0.2678	0.0035	0.0073
9	117.7	0.0	60.0	5720.	2521.	-17.	9647.	1197.	2003.
	43.9	0.0	49.8	5720.	2521.	-17.	9647.	1197.	2003.
	0.173		50.1	[illegible]	2697.	-18.	10322.	1270.	2143.
	90.	0.2221	50.1		57.46				
	681.1	0.9346	49.8	0.5995	0.3176	-0.0021	0.2313	0.0044	[illegible]
10	118.0	0.0	60.0	5915.	2647.	-3.	7452.	[illegible]	2194.
	44.0	0.0	49.9	5915.	2647.	-3.	7452.	[illegible]	2194.
	0.174		50.1	6324.	2832.	-3.	[illegible]	1155.	2347.
	90.	0.2221	49.9		60.10				
	681.1	0.9346	49.8	[illegible]	0.3326	-0.0004	[illegible]	0.0042	[illegible]
11	118.3	0.0	[illegible]	9255.	1989.	-99.	[illegible]	1017.	[illegible]
	44.2	0.0	50.0	9255.	1989.	-99.	[illegible]	1017.	[illegible]
	0.174		50.0	9922.	2111.	-105.	11043.	[illegible]	[illegible]
	91.	0.2217	50.0		44.55				
	681.7	0.9328	49.8	1.1572	[illegible]	[illegible]	[illegible]	[illegible]	[illegible]
12	117.8	0.0	50.0	[illegible]	2497.	-114.	10085.	[illegible]	[illegible]
	43.9	0.0	50.1	[illegible]	2497.	-114.	10085.	[illegible]	[illegible]
	0.173		50.1	[illegible]	2677.	-122.	11582.	[illegible]	[illegible]
	91.	[illegible]	50.2		58.43				
	681.7	[illegible]	49.9	[illegible]	0.3145	[illegible]	0.2585	[illegible]	[illegible]

TABLE V.— Continued.

RUN [illegible] TEST 525

PT	VKTS QPSF MTUN TEMP VSNDKTS	ALFSIL PSI RHOFLUG RHOR	THA PED POS LONG STK LAT STK PCH LEV	LIFT,O LIFT LIFT/W CL	DRAG,O DRAG DRAG/W FE CD	SIDE,O SIDE SIDE/W CY	PITCH,O PITCH PITCH/W CM	YAW,O YAW YAW/W CN	ROLL,O ROLL ROLL/W CROLL
13	110.[illegible]	0.0	50.0	5530.	2971.	+155.	14522.	190.	-569.
	93.9	0.0	50.0	5530.	2971.	+155.	14522.	190.	-569.
	0.174		50.0	9160.	3192.	+165.	15595.	204.	-740.
	92.	0.2213	50.0		67.63				
	682.3	0.9312	49.9	1.0728	0.3736	-0.0195	0.3475	0.0007	-0.0027
14	110.1	0.0	60.0	8265.	3191.	+84.	13468.	917.	+463.
	94.0	0.0	49.8	8265.	3191.	+84.	13468.	917.	+463.
	0.174		50.0	8876.	3427.	+90.	14463.	984.	+497.
	92.	0.2213	50.1		72.56				
	682.3	0.9312	49.8	1.03[illegible]	0.4039	-0.0106	0.3220	0.0036	-0.0018
15	1[illegible]8.0	0.0	40.0	7135.	3247.	+94.	13885.	415.	+329.
	93.8	0.0	49.9	7135.	3247.	+94.	13885.	415.	+329.
	0.173		50.1	7676.	3547.	+101.	14937.	447.	+353.
	93.	0.2209	50.0		75.25				
	683.9	0.9296	49.9	0.8997	0.4157	-0.0119	0.3332	0.0016	-0.0013
16	157.5	40.0	0.0	2135.	2048.	-85.	18422.	4143.	7434.
	76.9	0.0	50.0	2135.	2048.	-85.	18422.	4143.	7434.
	0.232		50.0	2332.	2237.	-93.	20004.	4520.	8007.
	95.	0.2176	49.9		26.65				
	684.2	0.9155	49.9	0.1555	0.1472	-0.0061	0.2546	0.0093	0.0177
17	157.7	0.0	0.0	7630.	1944.	-104.	13591.	4744.	8651.
	76.8	0.0	50.0	7630.	1944.	-109.	13591.	4709.	8651.
	0.231		50.1	8304.	2131.	-119.	14794.	5205.	9705.
	97.	0.2188	50.0		25.3[illegible]				
	685.9	0.9122	49.9	0.5926	0.1398	-0.0078	0.1860	0.0106	0.0198
18	155.2	4.0	0.0	13345.	2155.	-97.	5026.	3445.	7511.
	77.1	0.0	50.0	13345.	2155.	-97.	5026.	3445.	7511.
	0.232		50.1	14657.	2367.	-107.	5520.	3784.	8249.
	98.	0.2164	50.0		27.45				
	686.0	0.9105	49.8	0.9562	0.1544	-0.0070	0.0685	0.0077	0.0167
19	158.2	8.0	0.0	18457.	2732.	-83.	-2895.	1932.	8063.
	77.0	0.0	49.9	18457.	2732.	-83.	-2895.	1932.	8063.
	0.232		50.1	20311.	3006.	-91.	-3186.	2125.	8872.
	99.	0.2160	50.0		35.47				
	686.7	0.9089	49.9	1.3291	0.1960	-0.0060	-0.0388	0.0043	0.0180
20	158.4	10.0	0.0	20745.	3155.	+171.	+5293.	782.	3204.
	77.0	0.0	50.1	20745.	3155.	+171.	+5293.	782.	3204.
	0.232		50.1	22868.	3478.	+188.	+5774.	862.	3538.
	100.	0.2156	50.0		40.46				
	687.3	0.9072	49.7	1.4850	0.2263	+0.0123	+0.0716	0.0017	0.0072
21	158.6	-4.0	0.0	8310.	2239.	+130.	21907.	1895.	5907.
	76.9	0.0	50.0	8310.	2239.	+130.	21907.	1895.	5907.
	0.232		50.1	9196.	2476.	+144.	24249.	2097.	6538.
	102.	0.2147	50.0		29.10				
	688.5	0.9034	49.8	0.5967	0.1608	-0.0093	0.2999	0.0042	0.0132
22	158.5	0.0	0.0	13680.	2407.	-152.	18596.	2344.	3898.
	76.8	0.0	50.0	13680.	2407.	-152.	18596.	2344.	3898.
	0.232		50.1	15098.	2664.	-168.	20565.	2593.	4315.
	102.	0.2147	50.1		31.33				
	688.5	0.9035	49.8	0.9808	0.1731	-0.0109	0.2545	0.0052	0.0087
23	158.7	4.0	0.0	19195.	2880.	-225.	11733.	2620.	3423.
	76.9	0.0	49.9	19195.	2880.	-225.	11733.	2620.	3423.
	0.232		50.1	21229.	3186.	-249.	12911.	2883.	3795.
	103.	0.2144	50.0		37.53				
	689.1	0.9016	49.7	1.3755	0.2072	-0.0162	0.1604	0.0058	0.0076
24	158.8	8.0	0.0	23705.	3639.	-218.	938.	1407.	3378.
	76.9	0.0	49.9	23705.	3639.	-218.	938.	1407.	3378.
	0.232		50.1	26332.	4042.	-242.	1042.	1563.	3752.
	104.	0.2140	50.2		47.32				
	689.7	0.9002	49.7	1.7031	0.2615	-0.0157	0.0128	0.0031	0.0075

RUN [illegible] TEST 525

PT	VKTS QPSF MTIP TEMP VSNDKTS	ALFSIL PSI RHOFLUG RHOR	THA PED POS LONG STK LAT STK PCH LEV	LIFT,O LIFT LIFT/W CL	DRAG,O DRAG DRAG/W FE CD	SIDE,O SIDE SIDE/W CY	PITCH,O PITCH PITCH/W CM	YAW,O YAW YAW/W CN	ROLL,O ROLL ROLL/W CROLL
3	157.4	4.0	0.0	17415.	3024.	-156.	22081.	273.	-1267.
	76.9	0.0	50.1	17415.	3024.	-156.	22081.	273.	-1267.
	0.232		50.0	19527.	3314.	-171.	25104.	299.	-1390.
	97.	0.2166	49.9		39.34				
	685.4	0.9112	50.0	1.2204	0.2173	-0.0112	0.3129	0.0006	-0.0028
4	158.3	0.0	0.0	16495.	3233.	-187.	16511.	514.	-2680.
	76.9	0.0	50.0	16495.	3233.	-187.	16511.	514.	-2680.
	0.232		50.0	18144.	3567.	-206.	[illegible]	567.	-2946.
	107.	0.2154	50.0		42.05				
	687.3	0.9060	50.0	1.1850	0.2323	-0.0134	0.2532	[illegible]	[illegible]
5	158.7	4.0	0.0	21970.	3792.	-241.	8604.	1830.	-2245.
	77.1	0.0	50.2	21970.	3792.	-241.	8604.	1830.	-2245.
	0.232		50.0	24175.	4130.	-266.	[illegible]	2023.	-2481.
	10[illegible].	0.2151	49.9		44.52				
	687.9	0.9047	50.0	1.5665	0.2666	-0.0173	0.1126	0.0041	-0.0056
6	158.6	8.0	0.0	26195.	4107.	-244.	5402.	1451.	-1029.
	76.9	0.0	50.3	26195.	4107.	-244.	5403.	1451.	-1029.
	0.232		50.0	28741.	4548.	-267.	4321.	1607.	-1140.
	102.	0.2147	50.1		53.59				
	688.5	0.9031	50.1	1.7374	0.2956	-0.0173	0.0533	0.0052	-0.0023

TABLE V.— Continued.

RUN 42 TEST 525

PT	VEL QPSF [illegible] TEMP VSNDKTS	ALPHA PSI RHO[illegible] RHO	[illegible] PER POS LONG STA LAT STA PORT CTR	LIFT CL	DRAG CD	SIDE CY	PITCH CM	YAW CN	ROLL CRL
3	114.2	-4.0	50.0	5580.	2692.	-105.	15775.	544.	-450.
	42.7	0.0	50.0	5580.	2697.	-105.	15775.	544.	-450.
	0.171		50.0	5777.	2944.	-104.	16531.	564.	-443.
	71.	0.2246	50.0		67.76				
	669.2	0.9659	50.1	0.7223	0.3744	-0.0136	0.3544	0.0022	-0.0034
4	114.5	-2.0	50.0	6875.	2932.	-84.	15186.	546.	-1456.
	42.7	0.0	50.0	6875.	2932.	-89.	15187.	546.	-1456.
	0.171		49.9	7145.	3052.	-92.	16794.	622.	-1513.
	73.	0.2287	50.0		64.76				
	670.5	0.9622	50.1	0.8895	0.3870	-0.0115	0.3479	0.0024	-0.0159
5	114.3	-1.0	50.0	7230.	3057.	-77.	14242.	234.	-865.
	42.6	0.0	50.0	7230.	3057.	-77.	14242.	239.	-865.
	0.171		50.0	7514.	3177.	-80.	14801.	244.	-897.
	73.	0.2267	49.9		71.78				
	670.5	0.9623	50.2	0.9360	0.3966	-0.0100	0.3516	0.0013	-0.0235
6	115.7	-4.0	60.0	855.	2382.	-41.	14567.	1662.	1264.
	43.5	0.0	50.0	855.	2382.	-41.	14567.	1662.	1264.
	0.173		50.0	901.	2481.	-43.	15143.	1731.	1317.
	74.	0.2262	49.9		54.73				
	671.0	0.9601	50.1	0.1092	0.3023	-0.0052	0.3523	0.0066	0.0050
7	115.9	-2.0	50.0	2125.	2396.	-18.	13264.	1595.	2104.
	43.6	0.0	50.0	2125.	2396.	-18.	13264.	1595.	2104.
	0.173		50.0	2217.	2500.	-19.	13846.	1665.	2196.
	75.	0.2276	49.9		55.00				
	671.7	0.9583	50.1	0.2595	0.3039	-0.0023	0.3203	0.0063	0.0091
8	116.0	+1.0	50.0	2800.	2411.	-63.	12636.	1636.	1906.
	43.7	0.0	49.9	2800.	2411.	-63.	12636.	1638.	1906.
	0.173		50.0	2922.	2516.	-66.	13186.	1704.	1991.
	75.	0.2276	50.0		55.20				
	671.7	0.9583	50.1	0.3542	0.3050	-0.0080	0.3042	0.0064	0.0075
9	116.3	+2.0	50.0	7720.	1840.	-96.	12214.	612.	759.
	43.5	0.0	50.1	7720.	1840.	-96.	12214.	612.	759.
	0.173		50.0	8056.	1927.	-101.	12742.	641.	793.
	77.	0.2264	50.0		42.31				
	673.0	0.9548	50.1	0.9807	0.2337	-0.0122	0.2953	0.0029	0.0030
10	116.0	-2.0	50.0	7640.	2329.	-134.	12864.	746.	571.
	43.5	0.0	50.0	7640.	2329.	-139.	12864.	746.	571.
	0.173		50.0	8002.	2434.	-140.	13473.	781.	594.
	77.	0.2264	49.8		53.58				
	673.0	0.9548	50.1	0.9711	0.2960	-0.0177	0.3111	0.0024	0.0023
11	116.2	-2.0	50.0	7415.	2792.	-103.	14594.	1075.	149.
	43.6	0.0	50.1	7415.	2792.	-103.	14594.	1075.	149.
	0.173		50.0	7761.	2430.	-108.	15262.	1124.	156.
	78.	0.2265	50.0		64.10				
	673.6	0.9530	50.1	0.9473	0.3547	-0.0131	0.3510	0.0042	0.0006
12	115.9	-2.0	60.0	6960.	3026.	-56.	15122.	849.	-831.
	43.3	0.0	50.0	6960.	3026.	-56.	15122.	849.	-831.
	0.173		50.0	7324.	3175.	-59.	15806.	890.	-872.
	78.	0.2265	50.0		64.88				
	673.6	0.9531	50.1	0.8905	0.3861	-0.0064	0.3671	0.0034	-0.0033
13	116.2	-2.0	40.0	5005.	3176.	-106.	14476.	357.	-744.
	43.5	0.0	50.0	5005.	3176.	-106.	14476.	357.	-744.
	0.173		50.0	6315.	3339.	-111.	15743.	375.	-783.
	79.	0.2261	50.0		72.44				
	674.2	0.9512	50.2	0.7629	0.4033	-0.0135	0.3618	0.0016	-0.0024
14	98.9	-4.0	75.0	3770.	2305.	-55.	11195.	399.	-495.
	31.6	0.0	50.0	3770.	2305.	-55.	11195.	399.	-495.
	0.147		50.0	3954.	2418.	-58.	11742.	419.	-520.
	80.	0.2266	50.0		72.95				
	674.9	0.9534	50.2	0.6592	0.4030	-0.0096	0.3725	0.0022	-0.0065
15	99.1	-2.0	75.0	4650.	2360.	-71.	10925.	181.	-540.
	31.7	0.0	50.0	4650.	2360.	-71.	10925.	181.	-540.
	0.147		50.1	4877.	2475.	-74.	11459.	190.	-566.
	80.	0.2266	50.0		74.52				
	674.9	0.9534	50.3	0.8112	0.4117	-0.0124	0.3627	0.0010	-0.0029
16	99.1	0.0	75.0	5440.	2446.	-57.	10423.	612.	-91.
	31.7	0.0	50.0	5440.	2446.	-57.	10423.	612.	-91.
	0.147		50.2	5706.	2566.	-60.	10931.	642.	-96.
	80.	0.2266	50.0		77.24				
	674.9	0.9534	50.2	0.9491	0.4267	-0.0099	0.3127	0.0033	-0.0005
17	99.2	1.0	75.0	5900.	2473.	-74.	8998.	668.	115.
	31.8	0.0	50.0	5900.	2473.	-74.	8998.	668.	115.
	0.147		50.0	6189.	2594.	-82.	9436.	701.	121.
	80.	0.2266	50.0		[illegible]				
	674.9	0.9534	50.3	1.0263	0.4302	-0.0136	0.2979	0.0037	0.0006

TABLE V.— Continued.

RUN 43 TEST 525

PT	VKTS QPSF MACH TEMP VSNDFTS	ALPHA,C TEST RHO/RHO RHOF	INC PED POS LONG STK LAT STK PDL LEV	LIFT,C LIFT LIFT/Q CL	DRAG,C DRAG DRAG/Q FE CD	SIDE,C SIDE SIDE/Q CY	PITCH,C PITCH PITCH/Q CM	YAW,C YAW YAW/Q CN	ROLL,C ROLL ROLL/Q CROLL
4	115.5	-4.0	0.0	1115.	1093.	-56.	10352.	2365.	4020.
	43.8	0.0	50.1	1115.	1093.	-56.	10352.	2365.	4020.
	0.174		50.1	1169.	1146.	-59.	10850.	2474.	4213.
	77.	0.2266	50.0		24.43				
	673.0	0.9500	50.0	0.7405	0.1376	-0.0071	0.2483	0.0093	0.0157
5	118.0	0.0	0.0	4240.	1033.	-48.	6967.	2306.	4244.
	44.1	0.0	50.1	4240.	1033.	-48.	6967.	2306.	4244.
	0.174		50.0	4445.	1083.	-50.	7303.	2416.	4449.
	77.	0.2267	49.9		23.44				
	673.0	0.9500	50.0	0.5376	0.1295	-0.0060	0.1662	0.0090	0.0165
6	117.0	4.0	0.0	7255.	1135.	-42.	2854.	1802.	3786.
	44.0	0.0	50.0	7255.	1135.	-42.	2854.	1802.	3786.
	0.174		50.0	7644.	1196.	-47.	3008.	1899.	3991.
	80.	0.2255	50.0		25.86				
	674.9	0.9487	50.0	0.9110	0.1425	-0.0016	0.0662	0.0070	0.0146
7	117.1	8.0	0.0	10070.	1440.	-103.	-1866.	1104.	3820.
	44.0	0.0	50.0	10070.	1440.	-103.	-1866.	1104.	3820.
	0.174		50.0	10615.	1518.	-109.	-1966.	1164.	4027.
	80.	0.2255	49.9		32.72				
	674.9	0.9487	50.0	1.2638	0.1807	-0.0129	-0.0441	0.0043	0.0149
8	117.5	10.0	0.0	11575.	1639.	-116.	-2957.	931.	3072.
	44.0	0.0	50.2	11575.	1639.	-116.	-2957.	931.	3072.
	0.175		50.0	12203.	1728.	-122.	-3118.	982.	3239.
	80.	0.2255	50.1		36.44				
	674.9	0.9485	50.0	1.4413	0.2041	-0.0145	-0.0701	0.0036	0.0119
9	117.4	-4.0	0.0	4700.	1250.	-92.	11912.	1010.	2019.
	44.2	0.0	50.0	4700.	1250.	-92.	11912.	1010.	2019.
	0.174		50.0	4964.	1320.	-97.	12581.	1066.	2133.
	81.	0.2251	50.0		28.26				
	675.5	0.9468	49.9	0.5874	0.1552	-0.0115	0.2833	0.0039	0.0078
10	117.2	0.0	0.0	7800.	1337.	-104.	10277.	1403.	1992.
	44.0	0.0	50.0	7800.	1337.	-104.	10277.	1403.	1992.
	0.174		50.0	8237.	1412.	-110.	10853.	1481.	2104.
	81.	0.2251	49.9		30.35				
	675.5	0.9469	49.9	0.9784	0.1677	-0.0130	0.2453	0.0055	0.0078
11	117.5	4.0	0.0	10750.	1600.	-151.	6104.	1182.	1325.
	44.1	0.0	50.2	10750.	1600.	-151.	6104.	1182.	1325.
	0.174		50.0	11374.	1693.	-160.	6458.	1251.	1402.
	82.	0.2246	50.1		35.25				
	676.1	0.9457	50.0	1.3456	0.2003	-0.0189	0.1458	0.0046	0.0052
12	117.4	6.0	0.0	13425.	1987.	-146.	403.	777.	1556.
	44.1	0.0	50.1	13425.	1987.	-146.	403.	777.	1556.
	0.174		50.1	14204.	2102.	-154.	426.	823.	1647.
	82.	0.2247	50.0		45.09				
	676.1	0.9457	49.9	1.6831	0.2491	-0.0183	0.0096	0.0030	0.0061
13	117.5	8.0	0.0	13180.	1970.	-240.	1593.	373.	8505.
	44.1	0.0	50.1	13180.	1970.	-240.	1593.	373.	8505.
	0.174		49.9	13471.	2088.	-254.	1688.	396.	9016.
	83.	0.2242	49.1		44.86				
	676.7	0.9438	49.9	1.6506	0.2467	-0.0301	0.0380	0.0015	0.0331
14	117.5	9.0	0.0	12270.	1920.	-292.	3502.	-874.	18654.
	44.1	0.0	50.3	12270.	1920.	-292.	3502.	-874.	18654.
	0.174		49.9	13006.	2035.	-310.	3713.	-927.	19773.
	83.	0.2242	49.6		43.55				
	676.7	0.9439	49.9	1.5375	0.2406	-0.0366	0.0835	-0.0034	0.0727
15	117.4	-4.0	0.0	6635.	1706.	-85.	12192.	679.	1305.
	44.0	-10.0	50.1	6635.	1706.	-85.	12192.	679.	1305.
	0.174		50.0	7033.	1810.	-90.	12923.	720.	1384.
	83.	0.2242	50.1		38.78				
	676.7	0.9430	49.9	0.8323	0.2142	-0.0107	0.2910	0.0026	0.0051
16	117.5	-4.0	60.0	5760.	2932.	-104.	14126.	-178.	-482.
	44.0	0.0	50.1	5760.	2932.	-104.	14126.	-178.	-482.
	0.174		50.0	6117.	3114.	-110.	15003.	-189.	-511.
	84.	0.2236	50.0		66.57				
	677.0	0.9417	49.5	0.7225	0.3678	-0.0130	0.3372	-0.0007	-0.0019
17	117.6	0.0	60.0	8380.	3151.	-75.	12910.	274.	-546.
	44.0	0.0	50.1	8380.	3151.	-75.	12910.	274.	-546.
	0.174		50.1	8415.	3352.	-80.	13735.	292.	-581.
	85.	0.2234	49.9		71.57				
	678.0	0.9403	49.9	1.0522	0.3957	-0.0094	0.3065	0.0011	-0.0021
18	117.7	2.0	60.0	9665.	3314.	-123.	10514.	697.	-1324.
	44.1	0.0	50.1	9665.	3314.	-123.	10514.	697.	-1324.
	0.174		50.1	10281.	3531.	-131.	11505.	741.	-1408.
	85.	0.2234	49.9		75.31				
	678.0	0.9399	49.9	1.2092	0.4161	-0.0154	0.2580	0.0027	-0.0052
19	117.6	4.0	60.0	10715.	3537.	-36.	8885.	625.	139.
	44.0	0.0	50.1	10715.	3537.	-36.	8885.	625.	139.
	0.174		50.0	11399.	3763.	-38.	9219.	644.	148.
	85.	0.2234	49.9		80.39				
	678.0	0.9400	49.9	1.3451	0.4441	-0.0045	0.2071	0.0024	0.0005

TABLE V.— Continued.

RUN [illegible] TEST 525

PT	VEL S STDP MACH TEMP VSNTATS	ALPHA,C PSI RHUMIDU RHO	INA PEG POS LONG STA LAT STA PORT LEV	LIFT/C LIFT LIFT/C CL	DRAG/C DRAG DRAG/C FE CD	SIDE/C SIDE SIDE/C CY	PITCH/C PITCH PITCH/C CM	YAW/C YAW YAW/C CN	ROLL/C ROLL ROLL/C CROLL
5	117.4	4.0	0.0	1133.	1371.	-22.	11611.	2274.	3171.
	43.1	0.0	50.0	1133.	1371.	-22.	11611.	2274.	3171.
	0.174		50.0	1465.	1311.	-23.	11033.	2462.	3350.
	81.	0.2253	50.1		24.26				
	675.5	0.9456	50.1	0.1665	0.1341	-0.0025	0.2623	0.0045	0.0123
6	117.5	0.0	0.0	8233.	1034.	-24.	7185.	1627.	3923.
	44.2	0.0	49.9	8233.	1034.	-24.	7185.	1627.	3923.
	0.174		50.0	8477.	1094.	-25.	7605.	1934.	4152.
	82.	0.2246	50.0		23.44				
	676.1	0.9448	50.1	0.5249	0.1293	-0.0030	0.1709	0.0071	0.0152
7	117.5	4.0	0.0	7333.	1152.	-93.	2634.	2027.	3770.
	44.1	0.0	50.0	7333.	1152.	-93.	2634.	2027.	3770.
	0.174		49.4	7772.	1222.	-94.	2792.	2135.	3490.
	83.	0.2242	50.0		26.11				
	676.7	0.9431	50.1	0.9180	0.1443	-0.0116	0.0626	0.0074	0.0147
8	117.5	8.0	0.0	10325.	1434.	-127.	-1517.	1087.	2576.
	44.0	0.0	49.9	10325.	1434.	-127.	-1517.	1087.	2576.
	0.174		49.9	10427.	1521.	-133.	-1605.	1153.	2731.
	83.	0.2242	50.1		32.55				
	676.7	0.9431	50.0	1.2926	0.1799	-0.0114	-0.0362	0.0042	0.0100
9	117.6	10.0	0.0	11660.	1630.	-49.	-2713.	803.	2603.
	44.0	0.0	50.0	11660.	1630.	-49.	-2713.	803.	2603.
	0.174		50.0	12366.	1732.	-45.	-2882.	853.	2424.
	84.	0.2238	50.0		37.61				
	677.4	0.9412	50.1	1.4626	0.2035	-0.0112	-0.0648	0.0031	0.0104
10	117.7	4.0	0.0	4790.	1214.	-78.	12152.	1046.	1720.
	44.0	0.0	50.0	4790.	1214.	-78.	12152.	1046.	1720.
	0.174		50.0	5098.	1292.	-83.	12933.	1115.	1830.
	84.	0.2233	50.0		27.56				
	678.0	0.9396	50.0	0.6008	0.1523	-0.0098	0.2921	0.0041	0.0067
11	117.8	0.0	0.0	7650.	1315.	-103.	9983.	1285.	1879.
	44.1	0.0	50.0	7650.	1315.	-103.	9983.	1285.	1879.
	0.174		50.0	6355.	1400.	-110.	10624.	1368.	2000.
	85.	0.2233	50.0		29.01				
	678.0	0.9396	50.0	0.9531	0.1687	-0.0129	0.2379	0.0050	0.0073
12	117.8	4.0	0.0	10725.	1600.	-106.	5435.	1208.	1592.
	44.0	0.0	50.0	10725.	1600.	-106.	5435.	1208.	1592.
	0.174		50.0	11435.	1706.	-113.	5795.	1288.	1695.
	86.	0.2229	51.1		36.35				
	678.6	0.9379	50.0	1.3453	0.2007	-0.0133	0.1297	0.0047	0.0066
13	117.9	8.0	0.0	13595.	1972.	-145.	652.	538.	1274.
	44.1	0.0	50.0	13595.	1972.	-145.	652.	538.	1274.
	0.174		51.0	14445.	2103.	-155.	696.	574.	1356.
	86.	0.2229	50.0		44.56				
	678.6	0.9379	50.0	1.7017	0.2458	-0.0181	0.0155	0.0021	0.0050
14	117.8	8.0	0.0	13235.	1985.	-244.	548.	259.	8931.
	44.1	0.0	50.0	13235.	1986.	-244.	548.	259.	8931.
	0.174		49.7	14111.	2075.	-260.	650.	276.	9501.
	86.	0.2229	69.2		44.15				
	678.6	0.9379	50.0	1.6586	0.2838	-0.0306	0.0344	0.0010	0.0347
15	117.9	8.0	0.0	12345.	1890.	-289.	3773.	-593.	18317.
	45.1	0.0	69.9	12345.	1890.	-289.	3773.	-593.	18317.
	0.174		49.7	13163.	2015.	-308.	4023.	-633.	19531.
	86.	0.2224	89.3		42.82				
	678.6	0.9379	50.1	1.5852	0.2366	-0.0362	0.0899	-0.0023	0.0713
16	117.7	4.0	0.0	5390.	1635.	-88.	12309.	636.	331.
	44.9	0.0	50.0	5390.	1635.	-88.	12309.	636.	331.
	0.174		50.0	5825.	1760.	-94.	13142.	693.	357.
	87.	0.2225	49.8		37.24				
	679.2	0.9362	50.0	0.8047	0.2057	-0.0111	0.2986	0.0033	0.0013
17	116.2	0.0	0.0	9415.	1795.	-132.	9709.	941.	-2113.
	44.2	0.0	50.0	9415.	1795.	-132.	9709.	941.	-2113.
	0.174		49.4	10078.	1921.	-141.	10390.	1007.	-2261.
	88.	0.2221	49.4		46.82				
	679.9	0.9345	50.0	1.1775	0.2244	-0.0165	0.2310	0.0037	-0.0082
18	117.7	4.0	0.0	12390.	2097.	-124.	3085.	1230.	740.
	43.8	0.0	50.0	12390.	2097.	-124.	3085.	1230.	740.
	0.174		49.4	12457.	2244.	-133.	3443.	1291.	792.
	88.	0.2221	49.8		47.84				
	679.9	0.9346	50.0	1.5737	0.2643	-0.0156	0.0684	0.0047	0.0028
19	118.1	8.0	0.0	13560.	2265.	-125.	1167.	1207.	169.
	42.1	0.0	50.0	13560.	2265.	-125.	1167.	1207.	169.
	0.174		50.0	14511.	2445.	-134.	1249.	1292.	180.
	88.	0.2221	49.8		51.60				
	679.9	0.9345	50.0	1.5982	0.2862	-0.0157	0.0276	0.0047	0.0007
20	117.8	8.0	0.0	14565.	2511.	-162.	-2028.	811.	90.
	43.9	0.0	50.0	14565.	2511.	-162.	-2028.	811.	90.
	0.174		49.4	15593.	2687.	-173.	-2170.	867.	96.
	88.	0.2221	49.7		57.16				
	679.0	0.9345	50.0	1.8441	0.3158	-0.0204	-0.0465	0.0032	0.0003

TABLE V.– Continued.

RUN 45   TEST 525

PT	VELO / DESF / MION / TEMP / VSNDATS	ALPSIC / PSI / / MACH*100 / RHOF	YAW / FEED POS / LONG STA / LAT STA / PGM DEV	LIFT / LIFT / LIFT/Q / / CL	DRAG / DRAG / DRAG/Q / FE / CD	SIDE / SIDE / SIDE/Q / / CY	PITCH / PITCH / PITCH/Q / / CM	YAW / YAW / YAW/Q / / CN	ROLL / ROLL / ROLL/Q / / CROLL
3	119.7	-4.0	0.0	1320.	1130.	-29.	10917.	2606.	3363.
	47.4	0.0	50.3	13207.	1130.	-29.	10917.	2606.	3363.
	0.181		50.1	1352.	1157.	-30.	11176.	2672.	3444.
	65.	0.2321	50.0		23.66				
	665.9	0.9763	50.0	0.1520	0.1318	-0.0034	0.2423	0.0095	0.0122
4	120.1	0.0	0.0	4840.	1060.	-30.	7519.	1674.	2985.
	47.6	0.0	50.3	4840.	1060.	-30.	7519.	1674.	2985.
	0.181		50.1	4702.	1088.	-31.	7717.	1924.	3064.
	66.	0.2318	50.1		22.29				
	666.1	0.9722	50.0	0.5391	0.1232	-0.0035	0.1662	0.0066	0.0108
5	120.0	4.0	0.0	7725.	1197.	-98.	2304.	2020.	4032.
	47.6	0.0	50.0	7725.	1197.	-98.	2304.	2020.	4032.
	0.181		50.1	7943.	1231.	-101.	2369.	2077.	4145.
	67.	0.2312	50.0		25.26				
	666.7	0.9726	50.0	0.9006	0.1396	-0.0114	0.0511	0.0075	0.0146
6	119.9	8.0	0.0	11015.	1455.	-107.	-2407.	1550.	3711.
	47.3	0.0	50.2	11015.	1455.	-107.	-2407.	1550.	3711.
	0.180		50.1	11347.	1499.	-110.	-2535.	1597.	3823.
	66.	0.2307	50.0		30.76				
	667.3	0.9708	50.0	1.2873	0.1700	-0.0125	-0.0633	0.0056	0.0135
7	119.9	10.0	0.0	12325.	1672.	-149.	-5247.	1035.	2965.
	47.1	0.0	50.1	12325.	1672.	-149.	-5247.	1035.	2965.
	0.180		50.1	12719.	1725.	-154.	-5415.	1068.	3058.
	69.	0.2303	50.0		35.43				
	668.0	0.9693	50.0	1.4406	0.1957	-0.0175	-0.1170	0.0038	0.0108
8	119.8	12.0	0.0	12880.	1826.	-186.	-6562.	3.	1364.
	47.1	0.0	50.2	12880.	1826.	-186.	-6562.	3.	1364.
	0.180		50.2	13292.	1884.	-192.	-6772.	3.	1407.
	69.	0.2303	50.0		38.76				
	668.0	0.9690	50.1	1.5109	0.2141	-0.0218	-0.1464	0.0000	0.0050
9	119.5	14.0	0.0	6770.	1700.	-126.	12491.	1250.	-193.
	46.7	0.0	50.0	6770.	1700.	-126.	12491.	1250.	-193.
	0.179		50.1	6999.	1757.	-130.	12913.	1292.	-199.
	70.	0.2244	50.0		36.37				
	668.6	0.9673	50.0	0.8007	0.2009	-0.0149	0.2809	0.0046	-0.0007
10	119.6	0.0	0.0	9950.	1841.	-153.	8690.	1263.	-2005.
	46.7	0.0	50.1	9950.	1841.	-153.	8690.	1263.	-2005.
	0.179		50.2	10306.	1907.	-158.	9002.	1308.	-2077.
	71.	0.2245	50.0		39.40				
	669.2	0.9655	50.0	1.1766	0.2177	-0.0181	0.1955	0.0046	-0.0074
11	119.7	4.0	0.0	12965.	2135.	-146.	3601.	1306.	-1366.
	46.7	0.0	50.1	12965.	2135.	-146.	3601.	1306.	-1366.
	0.179		50.1	13454.	2216.	-152.	3744.	1355.	-1418.
	72.	0.2241	50.1		45.67				
	669.9	0.9637	50.0	1.5323	0.2523	-0.0173	0.0823	0.0048	-0.0050
12	119.7	6.0	0.0	14375.	2336.	-119.	143.	966.	-1609.
	46.7	0.0	50.1	14375.	2336.	-119.	143.	966.	-1609.
	0.179		50.0	14917.	2424.	-123.	149.	1002.	-1670.
	72.	0.2241	50.0		50.00				
	669.9	0.9637	50.0	1.6998	0.2762	-0.0141	0.0032	0.0035	-0.0059
13	119.5	7.0	0.0	14950.	2458.	-97.	-1530.	576.	-908.
	46.5	0.0	50.2	14950.	2458.	-97.	-1538.	576.	-908.
	0.179		50.2	15542.	2555.	-101.	-1599.	599.	-944.
	73.	0.2240	50.1		52.62				
	673.5	0.9615	50.2	1.7788	0.2918	-0.0115	-0.0348	0.0021	-0.0034
14	119.7	8.0	0.0	15500.	2600.	-166.	-2967.	781.	-1118.
	46.7	0.0	50.1	15500.	2600.	-166.	-2967.	781.	-1118.
	0.179		50.1	16114.	2707.	-193.	-3085.	812.	-1162.
	73.	0.2240	50.1		55.62				
	670.5	0.9619	50.1	1.6556	0.3084	-0.0220	-0.0669	0.0029	-0.0041
15	134.8	4.0	30.0	9005.	3062.	-167.	19955.	746.	-1242.
	52.6	0.0	50.2	9005.	3062.	-167.	19955.	746.	-1242.
	0.206		50.1	9455.	3225.	-176.	21062.	786.	-1309.
	77.	0.2257	50.0		44.49				
	673.0	0.9690	50.0	0.7903	0.2701	-0.0147	0.3556	0.0020	-0.0038
16	134.3	0.0	30.0	12810.	3395.	-163.	12951.	1025.	+353.
	62.2	0.0	50.1	12810.	3395.	-163.	12951.	1025.	+353.
	0.238		50.1	13517.	3582.	-172.	13665.	1085.	+373.
	78.	0.2253	50.0		54.55				
	673.6	0.9077	50.0	1.1372	0.3010	-0.0162	0.2166	0.0026	[illegible]
17	134.3	2.0	30.0	14730.	3669.	-216.	9654.	1394.	-1462.
	62.1	0.0	50.2	14730.	3669.	-216.	9654.	1394.	-1462.
	0.208		50.0	15571.	3875.	-228.	10165.	1474.	-1545.
	79.	0.2246	50.0		59.06				
	674.2	0.9460	49.9	1.3164	0.3269	-0.0194	0.0651	0.0039	[illegible]
18	140.0	4.0	30.0	16590.	4032.	-131.	5535.	468.	-759.
	62.7	0.0	50.2	16590.	4032.	-131.	5535.	468.	-759.
	0.209		50.2	17573.	4271.	-139.	5863.	510.	-804.
	83.	0.2246	50.0		64.33				
	674.9	0.9441	50.0	1.4823	0.3554	-0.0115	0.0926	0.0013	-0.0021
19	134.5	6.0	30.0	19615.	4717.	-168.	-1406.	-310.	-2440.
	62.1	0.0	50.3	19615.	4717.	-168.	-1406.	-310.	-2448.
	0.208		50.2	20817.	5005.	-178.	-1492.	-329.	-2597.
	81.	0.2240	49.9		75.45				
	675.5	0.9425	50.0	1.7450	0.4196	-0.0149	-0.0235	-0.0009	-0.0069
20	74.9	-4.0	90.0	2160.	1450.	-3.	6395.	-173.	-643.
	28.7	0.0	50.1	2160.	1450.	-3.	6395.	-173.	-643.
	0.124		50.1	2255.	1514.	-3.	6654.	-181.	-624.
	80.	0.2277	50.0		70.18				
	674.4	0.9561	50.0	0.5771	0.3874	-0.0008	0.3246	[illegible]	-0.0059

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OF POOR QUALITY

TABLE V.— Continued.

RUN 45 TEST 525

PT	VKTS QPSF MTON TEMP VSNDKTS	ALFS,C PSI RHO+IUG RHOP	INA PED POS LONG STA LAT STA POWLEV	LIFT,J LIFT LIFT/6 CL	DRAG,J DRAG DRAG/6 FE CD	SIDE,J SIDE SIDE/6 CY	PITCH,J PITCH PITCH/6 CM	YAW,J YAW YAW/6 CN	ROLL,J ROLL ROLL/6 CROLL
21	80.0	0.0	90.0	3390.	1522.	-4.	5502.	334.	-125.
	20.8	0.0	50.2	3390.	1522.	-4.	5502.	334.	-125.
	0.119		50.2	3539.	1589.	-4.	5744.	349.	-131.
	80.	0.2277	50.1		73.26				
	676.9	0.9579	49.9	0.7018	0.3398	-0.0001	0.2785	0.0128	-0.0010
22	79.8	4.0	90.0	4485.	1538.	0.	3723.	525.	-331.
	20.7	0.0	50.3	4485.	1538.	0.	3723.	525.	-331.
	0.119		50.1	4673.	1602.	0.	3880.	547.	-345.
	79.	0.2281	50.0		79.12				
	674.2	0.9597	50.0	1.1969	0.3371	0.0000	0.1891	0.0044	-0.0027
23	79.9	8.0	90.0	5575.	1916.	4.	1570.	569.	-72.
	20.8	0.0	50.1	5575.	1916.	4.	1570.	569.	-72.
	0.119		50.2	5809.	1992.	4.	1636.	593.	-75.
	79.	0.2281	50.0		87.52				
	674.2	0.9597	50.0	1.3843	0.4555	0.0011	0.0796	0.0047	-0.0008
24	79.9	10.0	90.0	5940.	1934.	-6.	495.	450.	-384.
	20.7	0.0	50.0	5940.	1934.	-6.	495.	450.	-384.
	0.119		50.1	6189.	2015.	-6.	516.	469.	-400.
	79.	0.2281	50.0		93.31				
	674.2	0.9597	50.0	1.5839	0.5155	-0.0016	0.0251	0.0037	-0.0032
25	79.9	12.0	90.0	6325.	2062.	-22.	-389.	61.	-491.
	20.8	0.0	50.0	6325.	2062.	-22.	-389.	61.	-491.
	0.119		50.1	6591.	2149.	-23.	-405.	63.	[illegible]
	79.	0.2281	50.0		94.57				
	674.2	0.9597	50.0	1.6881	0.5490	-0.0059	-0.0197	0.0015	-0.0082

RUN 46 TEST 525

PT	VKTS QPSF MTON TEMP VSNDKTS	ALFS,C PSI RHO+IUG RHOP	INA PED POS LONG STA LAT STA POWLEV	LIFT,J LIFT LIFT/6 CL	DRAG,J DRAG DRAG/6 FE CD	SIDE,J SIDE SIDE/6 CY	PITCH,J PITCH PITCH/6 CM	YAW,J YAW YAW/6 CN	ROLL,J ROLL ROLL/6 CROLL
5	116.9	-4.0	0.0	730.	1066.	-21.	10685.	1937.	2453.
	44.1	0.0	50.0	730.	1066.	-21.	10685.	1937.	2453.
	0.174		50.0	765.	1116.	-22.	11203.	2031.	2547.
	78.	0.2267	49.9		24.18				
	673.6	0.9538	50.0	0.0915	0.1336	-0.0026	0.2548	0.0075	0.0095
6	117.1	0.0	0.0	3440.	973.	-21.	6987.	1987.	2959.
	44.2	0.0	50.0	3440.	973.	-21.	6987.	1987.	2959.
	0.174		50.0	3614.	1022.	-22.	7340.	2087.	3108.
	79.	0.2263	49.7		22.02				
	674.2	0.9520	50.0	0.4301	0.1217	-0.0026	0.1663	0.0077	0.0115
7	117.1	4.0	0.0	6185.	1035.	-40.	3429.	1662.	2828.
	44.1	0.0	49.9	6185.	1035.	-40.	3429.	1662.	2828.
	0.174		50.0	6509.	1084.	-42.	3604.	1744.	2970.
	80.	0.2259	49.6		23.43				
	674.9	0.9502	50.0	0.7742	0.1296	-0.0050	0.0817	0.0065	0.0110
8	117.2	8.0	0.0	8695.	1276.	-57.	-1026.	459.	2477.
	44.1	0.0	50.0	8695.	1276.	-57.	-1026.	459.	2477.
	0.174		50.0	9167.	1345.	-60.	-1081.	[illegible]	2611.
	81.	0.2254	49.8		26.27				
	675.5	0.9485	49.9	1.0895	0.1596	-0.0071	-0.0245	0.0037	0.0046
9	117.0	10.0	0.0	9890.	1434.	-61.	-3331.	506.	3045.
	44.0	0.0	49.9	9890.	1434.	-61.	-3331.	506.	3045.
	0.174		49.9	10427.	1512.	-64.	-3512.	534.	3210.
	81.	0.2255	49.9		32.61				
	675.5	0.9485	50.0	1.2425	0.1802	-0.0077	-0.0796	0.0026	0.0119
10	117.3	-4.0	0.0	4835.	1375.	-95.	11287.	1144.	832.
	44.0	0.0	50.0	4835.	1375.	-95.	11287.	1144.	832.
	0.174		50.0	5010.	1455.	-101.	11944.	1210.	881.
	83.	0.2246	50.0		31.22				
	676.7	0.9457	49.9	0.6065	0.1725	-0.0119	0.2694	0.0045	0.0032
11	117.4	0.0	0.0	7635.	1493.	-102.	8593.	1071.	-177.
	44.1	0.0	50.0	7635.	1493.	-102.	8593.	1071.	-177.
	0.174		50.0	8044.	1577.	-105.	9046.	1134.	-187.
	81.	0.2245	49.9		32.74				
	676.7	0.9447	50.0	0.9823	0.1579	-0.0128	0.2053	0.0042	-0.0007
12	117.2	4.0	0.0	10556.	1707.	-134.	3725.	1036.	-494.
	44.0	0.0	49.9	10556.	1707.	-134.	3725.	1036.	-494.
	0.174		50.0	11167.	1801.	-142.	3943.	1094.	-521.
	83.	0.2246	49.4		38.84				
	676.7	0.9447	50.0	1.3267	0.2146	-0.0168	0.0891	0.0041	-0.0019
13	117.5	8.0	0.0	13237.	2085.	-134.	-2005.	732.	-83.
	44.1	0.0	49.9	13237.	2085.	-134.	-2005.	732.	-83.
	0.174		50.1	14031.	2211.	-142.	-2126.	777.	-88.
	84.	0.2241	49.4		47.31				
	677.4	0.9429	50.0	1.6586	0.2614	-0.0168	-0.0478	0.0029	-0.0003

TABLE V.— Continued.

RUN 67 TEST 525

PT	VKTS / QPSF / MTUN / TEMP / V5KOTS	ALPHA,C / PSI / RHOFIUN / RHOR	INA / FED POS / LONG STA / LAT STA / POS LEV	LIFT,U / LIFT / LIFT/B / CL	DRAG,U / DRAG / DRAG/B / FE / CD	SIDE,U / SIDE / SIDE/B / CY	PITCH,U / PITCH / PITCH/B / CM	YAW,U / YAW / YAW/B / CN	ROLL,U / ROLL / ROLL/B / CROLL
3	119.2	44.0	0.0	535.	1018.	-58.	10529.	2113.	4028.
	45.5	0.0	49.9	535.	1018.	-58.	10529.	2113.	4028.
	0.177		49.9	544.	1025.	-58.	11107.	2233.	4244.
	81.	0.2252	50.0		22.35				
	675.5	0.9474	49.9	0.0625	0.1235	-0.0044	0.2442	0.0080	0.0152
4	119.5	0.0	0.0	3595.	965.	-71.	7943.	2057.	2471.
	45.7	0.0	50.1	3595.	965.	-71.	7943.	2057.	2471.
	0.177		50.0	3749.	944.	-75.	8453.	2176.	2614.
	82.	0.2247	50.0		20.66				
	676.1	0.9456	49.9	0.4283	0.1142	-0.0086	0.1838	0.0077	0.0093
5	119.8	4.0	0.0	6680.	1005.	-63.	2476.	1634.	3161.
	45.9	0.0	50.0	6680.	1005.	-63.	2476.	1634.	3161.
	0.178		50.0	7084.	1065.	-67.	2625.	1732.	3351.
	81.	0.2242	50.1		21.93				
	676.7	0.9436	49.9	0.8036	0.1210	-0.0076	0.0566	0.0061	0.0118
6	119.7	8.0	0.0	9585.	1225.	-55.	-2138.	846.	2711.
	45.8	0.0	50.0	9585.	1225.	-55.	-2138.	846.	2711.
	0.176		50.0	10154.	1298.	-58.	-2265.	896.	2873.
	83.	0.2243	50.0		26.76				
	675.7	0.9435	50.0	1.1568	0.1478	-0.0066	-0.0488	0.0032	0.0102
7	119.9	10.0	0.0	13965.	1381.	-76.	-3995.	409.	2027.
	45.8	0.0	50.0	13965.	1381.	-76.	-3995.	409.	2027.
	0.176		50.0	14841.	1466.	-83.	-4242.	434.	2152.
	84.	0.2235	50.0		30.14				
	677.9	0.9417	50.0	1.6820	0.1665	-0.0094	-0.0917	0.0015	0.0076
8	119.5	14.0	0.0	4915.	1378.	-115.	12537.	1136.	818.
	45.4	0.0	50.0	4915.	1378.	-115.	12537.	1136.	818.
	0.177		50.0	5224.	1466.	-122.	13336.	1209.	870.
	85.	0.2235	49.9		30.33				
	678.0	0.9401	50.0	0.5977	0.1676	-0.0137	0.2401	0.0043	0.0031
9	119.6	0.0	0.0	8090.	1468.	-129.	9125.	1220.	-621.
	45.5	0.0	49.9	8090.	1468.	-129.	9125.	1220.	-621.
	0.177		50.0	8621.	1564.	-137.	9726.	1300.	-662.
	86.	0.2230	50.0		32.50				
	678.6	0.9384	49.9	0.9833	0.1784	-0.0157	0.2111	0.0036	-0.0023
10	119.6	4.0	0.0	11160.	1742.	-123.	4094.	794.	-125.
	45.5	0.0	50.0	11160.	1742.	-123.	4094.	794.	-125.
	0.177		49.9	11893.	1856.	-131.	4363.	846.	-134.
	86.	0.2230	50.0		35.32				
	678.6	0.9384	50.0	1.3569	0.2117	-0.0150	0.0937	0.0030	-0.0005
11	119.6	8.0	0.0	13925.	2097.	-144.	-1844.	811.	1328.
	45.8	0.0	50.0	13925.	2097.	-144.	-1844.	811.	1328.
	0.177		50.0	14839.	2235.	-153.	-1965.	865.	1415.
	86.	0.2231	50.0		46.18				
	679.6	0.9384	49.9	1.6942	0.2551	-0.0175	-0.0427	0.0031	0.0050

RUN 68 TEST 525

PT	VKTS / QPSF / MTUN / TEMP / V5KOTS	ALPHA,C / PSI / RHOFIUN / RHOR	INA / FED POS / LONG STA / LAT STA / POS LEV	LIFT,U / LIFT / LIFT/B / CL	DRAG,U / DRAG / DRAG/B / FE / CD	SIDE,U / SIDE / SIDE/B / CY	PITCH,U / PITCH / PITCH/B / CM	YAW,U / YAW / YAW/B / CN	ROLL,U / ROLL / ROLL/B / CROLL
3	78.0	44.0	0.0	2655.	634.	-14.	5652.	234.	417.
	22.5	0.0	50.0	2655.	639.	-14.	5652.	234.	417.
	0.118		49.9	2671.	634.	-14.	5697.	236.	424.
	62.	0.2359	50.2		30.44				
	663.5	0.9920	50.0	0.5599	0.1712	-0.0038	0.3607	0.0028	0.0077
4	78.1	44.0	60.0	1990.	1345.	-39.	7983.	152.	59.
	20.5	0.0	50.0	1990.	1345.	-39.	7983.	152.	59.
	0.118		49.9	1955.	1355.	-39.	8044.	153.	59.
	62.	0.2359	50.3		65.66				
	663.5	0.9920	50.0	0.5232	0.3627	-0.0105	0.4097	0.0013	0.0005
5	79.1	44.0	90.0	1455.	1439.	-27.	7199.	541.	155.
	20.5	0.0	49.9	1455.	1439.	-27.	7199.	541.	155.
	0.118		50.0	1466.	1450.	-27.	7254.	546.	157.
	62.	0.2359	50.1		70.24				
	663.5	0.9920	50.0	0.3924	0.3881	-0.0073	0.3695	0.0050	0.0015
6	115.6	44.0	60.0	6565.	2983.	-88.	17440.	7.	55.
	40.9	0.0	50.0	6565.	2983.	-88.	17440.	7.	55.
	0.176		49.8	4477.	3055.	-90.	16370.	7.	56.
	56.	0.2321	50.2		66.44				
	666.1	0.9766	49.9	0.5358	0.3664	-0.0108	0.4194	0.0001	0.0002
7	115.7	42.0	60.0	5635.	3055.	-85.	15794.	958.	175.
	40.9	0.0	50.0	5635.	3055.	-85.	15794.	958.	175.
	0.176		44.9	5781.	3134.	-87.	16204.	983.	179.
	57.	0.2317	50.2		[illegible]				
	666.7	0.9767	50.0	0.6930	0.3757	-0.0105	0.3697	0.0037	0.0007
8	116.6	4.0	60.0	6515.	3137.	-108.	14244.	1053.	262.
	40.9	0.0	50.0	6515.	3137.	-108.	14244.	1053.	262.
	0.176		49.9	6842.	3216.	-111.	14611.	1080.	269.
	57.	0.2317	50.1		94.40				
	666.7	0.9764	49.9	0.8398	0.3862	-0.0133	[illegible]	[illegible]	[illegible]

TABLE V.— Continued.

RUN 48 TEST 525

PT	MPH QPSF MACH TEMP RVSNORTS	ALPHA PSI [illegible] [illegible]	[illegible] FIN POST [illegible] LAT [illegible] POWERCENT	LIFT LIFT LIFT/Q CL	DRAG DRAG DRAG/Q KE CD	SIDE SIDE SIDE/Q CY	PITCH PITCH PITCH/Q CM	YAW YAW YAW/Q CN	ROLL ROLL ROLL/Q CROLL
9	116.5	2.0	60.0	8170.	3255.	+119.	11638.	1031.	-574.
	44.9	0.0	50.0	8170.	3255.	-119.	11638.	1031.	-574.
	0.176		50.0	8346.	3348.	-122.	11902.	1054.	-595.
	68.	0.2312	50.2		72.40				
	667.3	0.9729	50.0	0.9986	0.4033	-0.0146	0.2724	0.0039	-0.0022
10	116.8	4.0	60.0	9275.	3418.	-61.	9306.	1540.	364.
	44.8	0.0	49.9	9275.	3418.	-61.	9306.	1540.	364.
	0.175		50.0	9351.	3520.	-65.	9565.	1590.	380.
	69.	0.2308	50.2		76.24				
	668.0	0.9711	50.0	1.1430	0.4212	-0.0075	0.2182	0.0051	0.0014
11	116.7	5.0	60.0	9895.	3510.	-67.	7598.	1363.	491.
	44.8	0.0	50.0	9895.	3510.	-67.	7598.	1363.	491.
	0.175		50.0	10190.	3615.	-69.	7824.	1404.	505.
	69.	0.2306	50.2		78.33				
	668.3	0.9711	49.9	1.2200	0.4328	-0.0083	0.1783	0.0052	0.0019
12	116.8	6.0	60.0	10465.	3579.	-55.	6144.	1320.	150.
	44.7	0.0	50.0	10465.	3579.	-55.	6144.	1320.	150.
	0.175		50.0	10797.	3692.	-57.	6338.	1362.	155.
	70.	0.2304	50.2		80.00				
	668.6	0.9693	50.0	1.2921	0.4420	-0.0068	0.1444	0.0051	0.0006
13	116.6	7.0	60.0	11085.	3700.	-71.	4419.	1421.	215.
	44.6	0.0	49.9	11085.	3700.	-71.	4419.	1421.	215.
	0.175		49.9	11395.	3811.	-73.	4559.	1466.	223.
	70.	0.2304	50.2		82.96				
	668.6	0.9693	50.0	1.3652	0.4583	-0.0088	0.1042	0.0055	0.0008
14	116.6	8.0	60.0	11510.	3778.	-62.	3017.	1340.	648.
	44.5	0.0	50.0	11510.	3778.	-62.	3017.	1340.	648.
	0.175		49.9	11896.	3905.	-64.	3118.	1385.	677.
	71.	0.2301	50.2		84.88				
	669.2	0.9675	50.0	1.4287	0.4690	-0.0077	0.0713	0.0052	0.0033
15	116.4	9.0	60.0	12075.	3871.	-91.	1629.	1262.	-454.
	44.4	0.0	49.9	12075.	3871.	-91.	1629.	1262.	-454.
	0.175		49.9	12480.	4001.	-94.	1684.	1305.	-469.
	71.	0.2300	50.2		87.15				
	669.2	0.9676	50.0	1.5029	0.4815	-0.0113	0.0386	0.0049	-0.0018
16	116.4	10.0	60.0	12375.	4001.	-125.	539.	942.	-548.
	44.2	0.0	49.8	12375.	4001.	-125.	539.	942.	-548.
	0.174		50.0	12837.	4150.	-130.	559.	1024.	-620.
	73.	0.2291	50.2		90.55				
	670.5	0.9640	50.0	1.5074	0.5003	-0.0156	0.0128	0.0039	-0.0023
17	117.0	6.0	60.0	10370.	3591.	-1006.	7103.	7960.	-2441.
	44.6	-6.0	50.0	10370.	3591.	-1006.	7103.	7960.	-2441.
	0.175		49.4	10779.	3733.	-1046.	7384.	8274.	-2551.
	74.	0.2287	50.2		80.51				
	671.1	0.9627	49.9	1.2848	0.4448	-0.1246	0.1679	0.0327	-0.0096
18	117.3	6.0	60.0	10530.	3595.	-125.	7111.	1606.	-506.
	44.7	0.0	50.0	10530.	3595.	-125.	7111.	1606.	-506.
	0.175		50.0	10966.	3754.	-130.	7406.	1734.	-527.
	75.	0.2282	50.0		80.15				
	671.7	0.9602	50.0	1.3003	0.4027	-0.0150	0.1671	0.0064	-0.0019
19	117.6	6.0	60.0	10450.	3618.	772.	6731.	-5010.	1513.
	44.9	4.0	50.0	10450.	3618.	772.	6731.	-5010.	1513.
	0.176		50.1	10863.	3764.	804.	7013.	-5218.	1576.
	75.	0.2282	50.1		80.41				
	671.7	0.9601	50.0	1.2820	0.4002	0.0949	0.1574	-0.0191	0.0058
20	117.4	8.0	60.0	10175.	3603.	1529.	7808.	-8486.	2370.
	44.6	8.0	50.0	10175.	3603.	1529.	7808.	-8486.	2370.
	0.175		50.0	10636.	3608.	1598.	8246.	-8870.	2477.
	77.	0.2274	50.1		81.60				
	673.0	0.9567	50.0	1.2591	0.4500	0.1892	0.1858	-0.0326	0.0041
21	117.6	6.0	60.0	10345.	3653.	-1324.	8250.	10073.	-2560.
	44.7	-6.0	50.0	10345.	3553.	-1324.	8250.	10073.	-2560.
	0.175		50.0	10834.	3820.	-1387.	8640.	10544.	-2792.
	78.	0.2270	50.2		81.78				
	673.6	0.9579	50.0	1.2795	0.4518	-0.1638	0.1942	-0.0387	-0.0103
22	120.3	8.0	0.0	12485.	2154.	-936.	4180.	5587.	-6486.
	46.7	-6.0	50.0	12485.	2154.	-936.	4180.	5587.	-6486.
	0.179		50.0	13104.	2262.	-931.	4345.	5866.	-6791.
	79.	0.2264	50.1		40.12				
	674.2	0.9524	50.1	1.4771	0.2548	-0.1072	0.0942	0.0205	-0.0238
23	120.4	8.0	0.0	12575.	2112.	-163.	2404.	335.	-2449.
	46.6	0.0	49.9	12575.	2112.	-163.	2404.	335.	-2449.
	0.179		50.0	13.54.	2222.	-170.	2592.	352.	-2625.
	80.	0.2260	50.2		45.29				
	674.9	0.9507	49.9	1.4818	0.2502	-0.0193	0.0555	0.0012	-0.0097
24	120.5	8.0	0.0	12565.	2113.	690.	5011.	-2302.	6724.
	46.7	0.0	49.9	12565.	2113.	690.	5011.	-2302.	6724.
	0.179		50.0	13217.	2225.	726.	5187.	-2421.	7073.
	80.	0.2260	50.1		45.25				
	674.9	0.9507	50.0	1.4865	0.2500	0.0816	0.0870	-0.0085	0.0247
25	120.2	0.0	0.0	12230.	2168.	1412.	5743.	-7695.	4755.
	46.5	8.0	50.1	12230.	2168.	1412.	5743.	-7695.	4755.
	0.179		50.0	12864.	2280.	1485.	6041.	-8095.	5002.
	80.	0.2258	50.2		45.54				
	674.9	0.9507	50.0	1.4519	0.2574	0.1676	0.1284	-0.0284	0.0350
26	120.2	8.0	0.0	11705.	2255.	+18194.	5221.	+12501.	11860.
	46.6	12.0	50.0	11705.	2255.	+18194.	5221.	+12501.	11860.
	0.179		50.0	12334.	2370.	+4019.	4716.	+22432.	11025.
	81.	0.2250	50.2		45.55				
	675.5	0.9490	50.0	1.3923	0.2682	-0.5396	0.2097	-0.1.2.	-0.1102

TABLE V.— Continued.

RUN [illegible] TEST 525

PT	VKTS QPSF MTUN TEMP VSNDKTS	ALFS.C PSI MHC*100 AMDR	INA PED POS LONG STK LAT STK RUD LEV	LIFT.U LIFT LIFT/6 CL	DRAG.U DRAG DRAG/6 FE CD	SIDE.U SIDE SIDE/6 CY	PITCH.U PITCH PITCH/6 CM	YAW.U YAW YAW/6 CN	ROLL.U ROLL ROLL/6 CROLL
3	116.5	-4.0	60.0	4985.	1435.	-135.	12479.	720.	490.
	45.0	0.0	50.0	4985.	1435.	-135.	12479.	720.	490.
	0.176		50.1	5288.	1465.	-139.	13144.	755.	500.
	63.	0.2124	50.0		31.69				
	664.2	0.9799	50.1	0.5121	0.1782	-0.0166	0.3019	0.0027	0.0019
4	116.7	-4.0	60.0	4920.	3104.	-155.	17966.	739.	-489.
	45.0	0.0	50.0	4920.	3104.	-155.	17966.	739.	-489.
	0.176		50.0	5040.	3162.	-159.	18325.	757.	-501.
	65.	0.2320	50.1		69.62				
	665.2	0.9761	50.1	0.6044	0.3813	-0.0190	[illegible]	0.0028	[illegible]
5	116.9	0.0	60.0	7615.	3278.	-74.	13240.	987.	-484.
	45.0	0.0	50.0	7615.	3278.	-74.	13240.	987.	-484.
	0.176		50.1	7815.	3365.	-76.	13593.	1013.	-497.
	66.	0.2316	50.1		72.70				
	666.1	0.9792	50.1	0.9391	0.4021	-0.0091	0.1591	0.0058	-0.0014
6	116.9	4.0	60.0	10205.	3614.	-65.	7087.	1031.	-287.
	45.0	0.0	50.0	10205.	3614.	-65.	7087.	1031.	-287.
	0.176		50.1	10495.	3716.	-67.	7288.	1061.	-295.
	67.	0.2311	50.1		68.32				
	666.7	0.9720	50.1	1.2531	0.4438	-0.0080	0.0656	0.0039	-0.0011
7	117.1	8.0	60.0	11545.	3624.	-124.	3496.	1264.	-745.
	45.0	0.0	50.2	11545.	3624.	-124.	3496.	1264.	-745.
	0.176		50.0	11895.	3946.	-115.	4017.	1251.	-765.
	65.	0.2307	49.9		84.94				
	667.3	0.9705	50.1	1.4182	0.4691	-0.0158	0.0933	0.0046	-0.0026
8	117.0	8.0	60.0	12700.	4045.	-169.	985.	1090.	-575.
	44.9	0.0	50.1	12700.	4045.	-167.	985.	1090.	-575.
	0.176		50.1	13100.	4177.	-174.	1020.	1125.	-1006.
	64.	0.2303	50.1		90.06				
	668.0	0.9688	50.0	1.5619	0.4976	-0.0208	0.0206	0.0042	-0.0037

RUN 50 TEST 525

PT	VKTS QPSF MTUN TEMP VSNDKTS	ALFS.C PSI MHC*100 AMDR	INA PED POS LONG STK LAT STK RUD LEV	LIFT.U LIFT LIFT/6 CL	DRAG.U DRAG DRAG/6 FE CD	SIDE.U SIDE SIDE/6 CY	PITCH.U PITCH PITCH/6 CM	YAW.U YAW YAW/6 CN	ROLL.U ROLL ROLL/6 CROLL
3	117.3	-4.0	60.0	4665.	3077.	-136.	18112.	557.	-510.
	45.0	0.0	50.0	4665.	3077.	-136.	18112.	557.	-510.
	0.176		49.9	4834.	3188	-141.	18767.	577.	-528.
	71.	0.2294	50.1		68.42				
	669.2	0.9651	50.0	0.5731	0.3780	-0.0167	0.4234	0.0021	-0.0019
4	117.2	0.0	60.0	7130.	3255.	-138.	13521.	966.	-381.
	44.8	0.0	50.0	7130.	3255.	-138.	13521.	966.	-381.
	0.176		50.0	7401.	3379	-143.	14036.	1005.	-396.
	72.	0.2290	50.2		72.64				
	669.9	0.9633	50.1	0.8191	0.4013	-0.0170	0.3172	0.0037	-0.0015
5	117.5	4.0	60.0	9725.	3544.	-101.	7777.	871.	-838.
	44.9	0.0	49.9	9725.	3544.	-101.	7777.	871.	-838.
	0.176		50.1	10115.	3686	-105.	8089.	906.	-872.
	73.	0.2285	50.1		78.93				
	670.5	0.9615	49.9	1.1966	0.4361	-0.0124	0.182.	0.0033	-0.0032
6	117.7	6.0	60.0	10900.	3732.	-97.	5172.	1379.	-238.
	45.0	0.0	50.0	10900.	3732.	-97.	5172.	1379.	-238.
	0.176		49.9	11358.	3889	-101.	5390.	1437.	-248.
	74.	0.2281	50.0		82.99				
	671.1	0.9597	50.0	1.3391	0.4585	-0.0119	0.1209	0.0053	-0.0009
7	117.8	8.0	60.0	11960.	3941.	-99.	2161.	1593.	-652.
	45.0	0.0	50.0	11960.	3941.	-99.	2161.	1593.	-652.
	0.176		50.0	12507.	4114	-103.	2277	1663.	-681.
	75.	0.2277	50.2		87.59				
	671.7	0.9579	50.0	1.4711	0.4839	-0.0122	0.0510	0.0061	-0.0025
8	117.9	6.0	60.0	10585.	3859.	-1761.	7767.	11890.	-4531.
	45.0	-8.0	50.1	10585.	3859.	-1761.	7767.	11890.	-4531.
	0.176		49.9	11071.	4036	-1842.	8123.	12436.	-4739.
	76.	0.2273	50.5		85.81				
	672.4	0.9561	50.0	1.3004	0.4741	-0.2163	0.1616	0.0454	-0.0173
9	118.0	6.0	60.0	10760.	3762.	-972.	6692.	7622.	-2760.
	45.0	-4.0	50.0	10760.	3762.	-972.	6692.	7622.	-2760.
	0.176		50.0	11275.	3942	-1019.	7222.	8197.	-2913.
	77.	0.2268	50.0		83.65				
	673.0	0.9543	50.0	1.3219	0.4622	-0.1194	0.1161	0.0299	-0.0106
10	117.9	6.0	60.0	10965.	3712.	-87.	5910.	1375.	-266.
	44.9	0.0	50.0	10965.	3712.	-87.	5910.	1375.	-266.
	0.176		50.0	11490.	3890	-91.	6193.	1440.	-278.
	77.	0.2266	50.2		82.53				
	673.0	0.9545	50.0	1.3485	0.4565	-0.0107	0.1563	0.0053	-0.0010
11	117.9	6.0	60.0	10750.	3703.	746.	6664.	-4022.	1145.
	44.9	4.0	50.0	10750.	3703.	746.	6664.	-4022.	1145.
	0.176		50.0	11264.	3852	764.	7504	-4215.	1200.
	77.	0.2268	50.1		82.51				
	673.0	0.9543	50.0	1.3234	0.4559	0.0921	0.1566	-0.0154	0.0044
12	117.9	6.0	60.0	10555.	3803.	1526.	8336.	-6355.	1781.
	44.8	8.0	50.0	10555.	3803.	1526.	8336.	-6355.	1781.
	0.176		50.1	11084.	3991	1605.	8756.	-6774.	1870.
	78.	0.2263	50.2		64.76				
	673.8	0.9522	50.0	1.3307	0.4683	0.1883	0.1955	-0.0324	0.0066
13	118.2	6.0	90.0	9905.	3853.	-33.	7305.	1412.	980.
	44.9	0.0	49.6	9905.	3853.	-33.	7305.	1412.	980.
	0.176		50.0	10421.	4031	-35.	7686.	1486.	1031.
	79.	0.2259	50.1		65.70				
	674.2	0.9504	50.0	1.2161	0.4735	-0.0041	0.1710	0.0054	0.0037

TABLE V.– Concluded.

RUN 51 TEST 525

PT	VARS QPSF MTUN TEMP VSNDKTS	ALFS,C PSI RHO*100 RHOR	INA PED POS LONG STK LAT STK POW LEV	LIFT,U LIFT LIFT/B CL	DRAG,U DRAG DRAG/B FE CD	SIDE,U SIDE SIDE/B CY	PITCH,U PITCH PITCH/B CM	YAW,U YAW YAW/B CN	ROLL,U ROLL ROLL/B CROLL
3	118.2 44.9 0.176 78. 673.6	-4.0 0.0 0.2260 0.9509	0.0 50.0 50.0 50.0 50.0	-113. -113. -116. -0.0135	803. 803. 844 17.87 0.0987	-65. -65. -68. -0.0080	9575. 9575. 14069. 0.2240	1883. 1883. 1981. 0.0672	-500. -500. -526. -0.0019
4	118.4 45.0 0.176 80. 674.9	-2.0 0.0 0.2252 0.9474	0.0 50.0 50.0 50.0 50.0	1480. 1480. 1562. 0.1818	762. 762. 804 16.94 0.0936	-45. -45. -47. -0.0055	7380. 7380. 7793. 0.1725	1877. 1877. 1981. 0.0072	-45. -45. -48. -0.0002
5	118.6 44.9 0.176 82. 676.1	0.0 0.0 0.2244 0.9439	0.0 50.0 50.0 50.0 50.0	3050. 3050. 3231. 0.3749	734. 734. 778 16.33 0.0902	-59. -59. -63. -0.0073	5846. 5846. 6196. 0.1369	1865. 1865. 1976. 0.0071	-390. -390. -413. -0.0015
6	118.6 45.0 0.176 82. 676.1	2.0 0.0 0.2244 0.9439	0.0 50.0 50.0 50.0 50.0	4570. 4570. 4842. 0.5614	754. 754. 799 16.77 0.0926	-91. -91. -96. -0.0112	3317. 3317. 3514. 0.0775	1800. 1800. 1907. 0.0069	-577. -577. -612. -0.0022
7	118.5 44.9 0.176 82. 676.1	4.0 0.0 0.2244 0.9439	0.0 50.0 50.0 50.0 50.0	5990. 5990. 6346. 0.7370	816. 816. 864 18.18 0.1005	-78. -78. -83. -0.0096	1656. 1656. 1755. 0.0388	1543. 1543. 1635. 0.0059	-1366. -1366. -1447. -0.0052
8	119.0 45.1 0.176 83. 676.7	6.0 0.0 0.2239 0.9421	0.0 50.0 50.0 50.0 50.0	7400. 7400. 7855. 0.9059	917. 917. 973 20.32 0.1123	-86. -86. -91. -0.0105	-1401. -1401. -1487. -0.0326	1707. 1707. 1812. 0.0065	-309. -309. -328. -0.0012
9	118.8 44.9 0.176 84. 677.4	8.0 0.0 0.2235 0.9404	0.0 50.0 50.0 50.0 50.0	8490. 8490. 9028. 1.0441	1048. 1048. 1114 23.33 0.1289	-100. -100. -106. -0.0123	-3293. -3293. -3502. -0.0771	1555. 1555. 1651. 0.0059	410. 410. 435. 0.0016
10	118.8 44.9 0.176 84. 677.4	10.0 0.0 0.2235 0.9404	0.0 50.0 50.0 50.0 50.0	9365. 9365. 9958. 1.1513	1219. 1219. 1296 27.12 0.1498	-98. -98. -104. -0.0122	-6201. -6201. -6594. -0.1450	1688. 1688. 1794. 0.0066	-1339. -1339. -1424. -0.0053

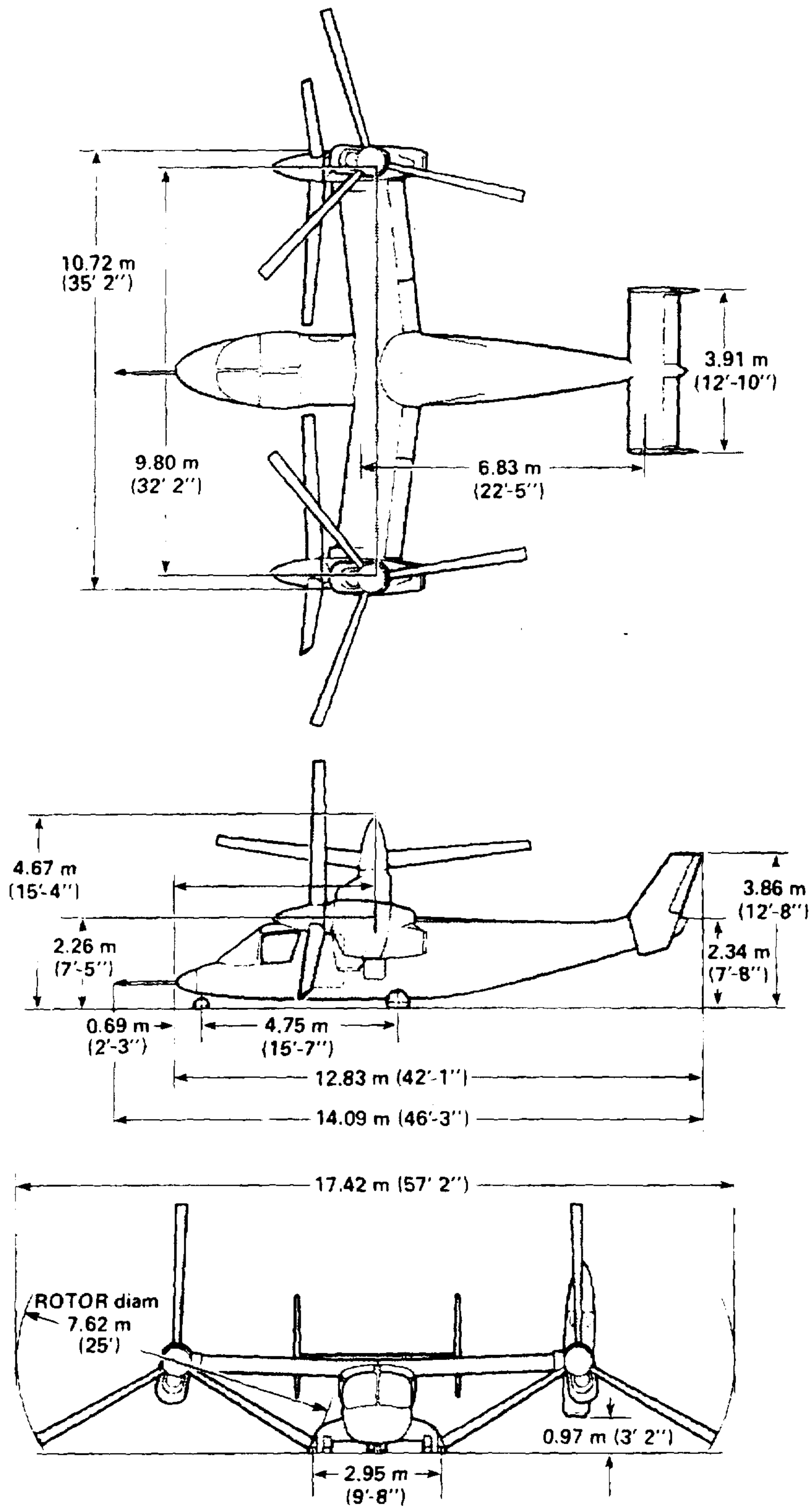

Figure 1.- Aircraft geometry.

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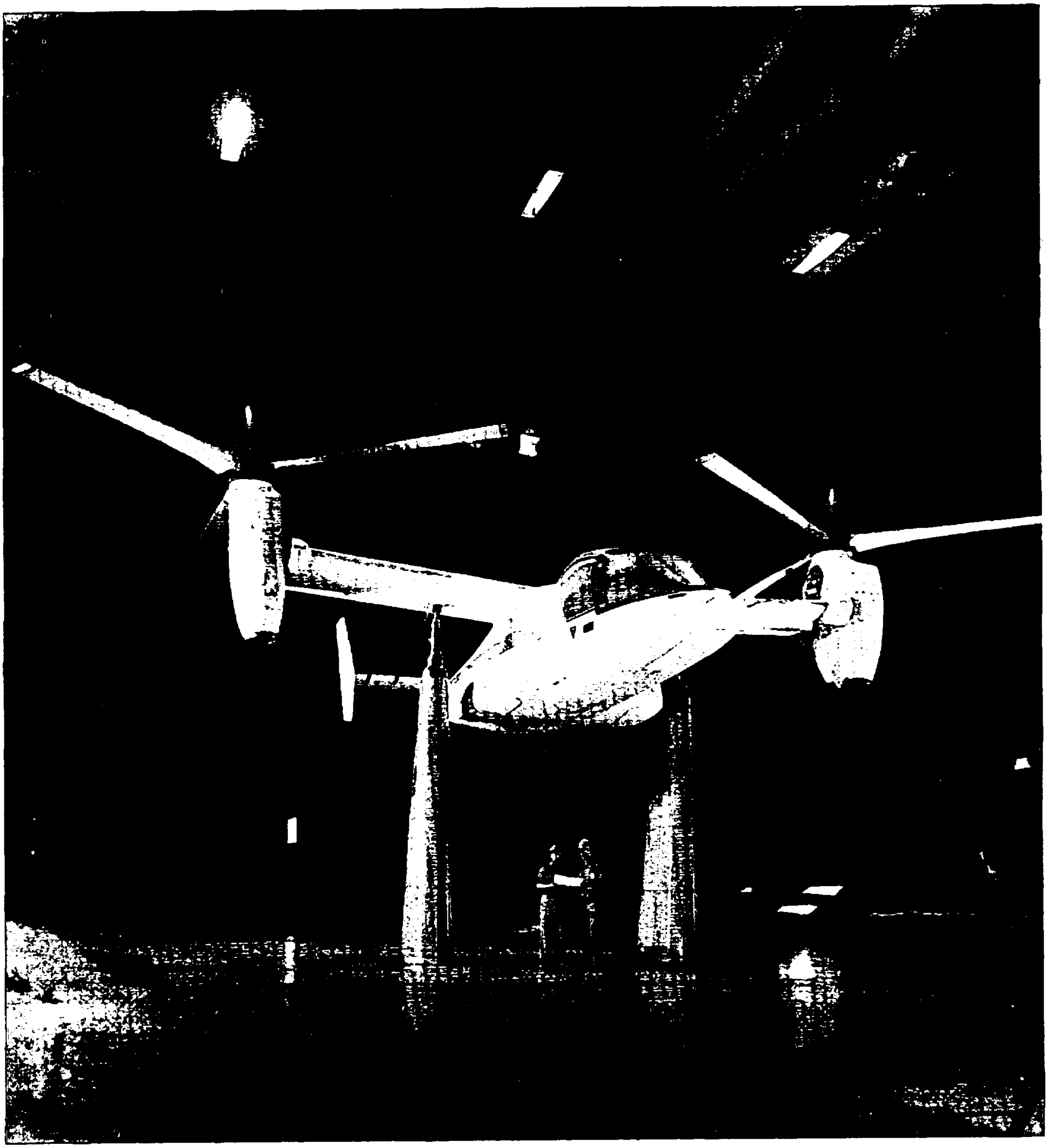

(a) Helicopter mode.

Figure 2.— Aircraft mounted in the tunnel.

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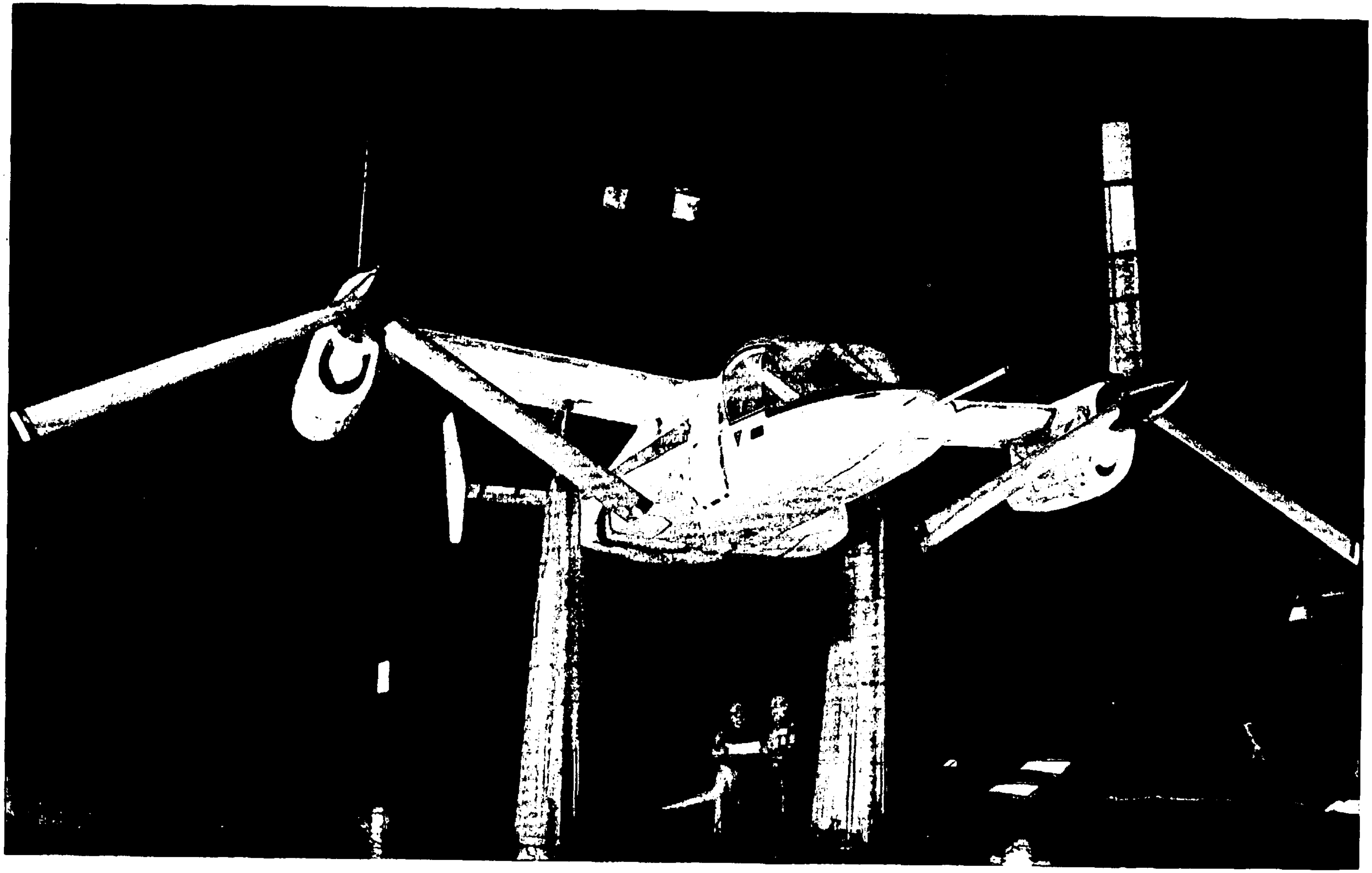

(b) Airplane mode.

Figure 2.- Concluded.

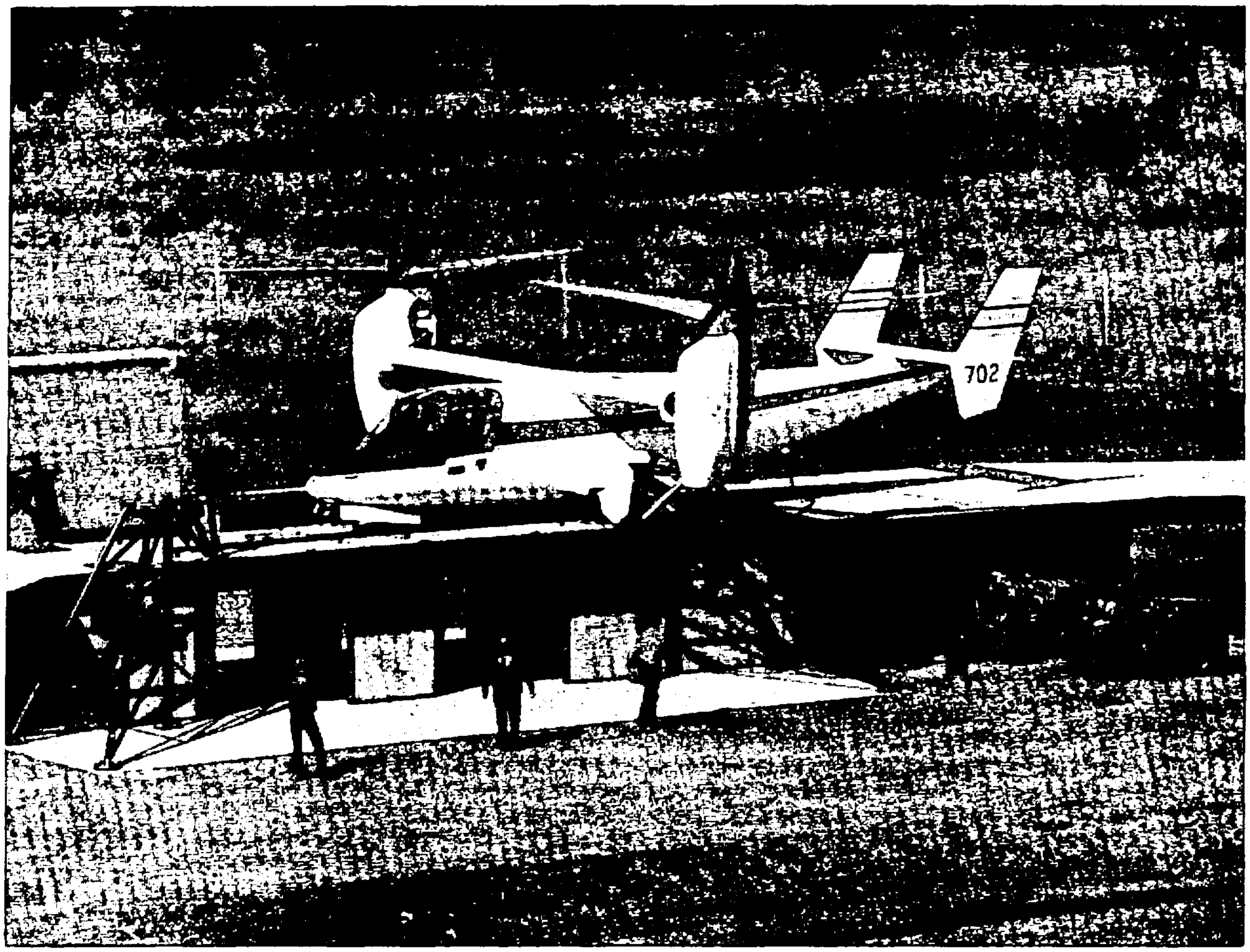

Figure 3.- Aircraft on the ground tie-down stand.

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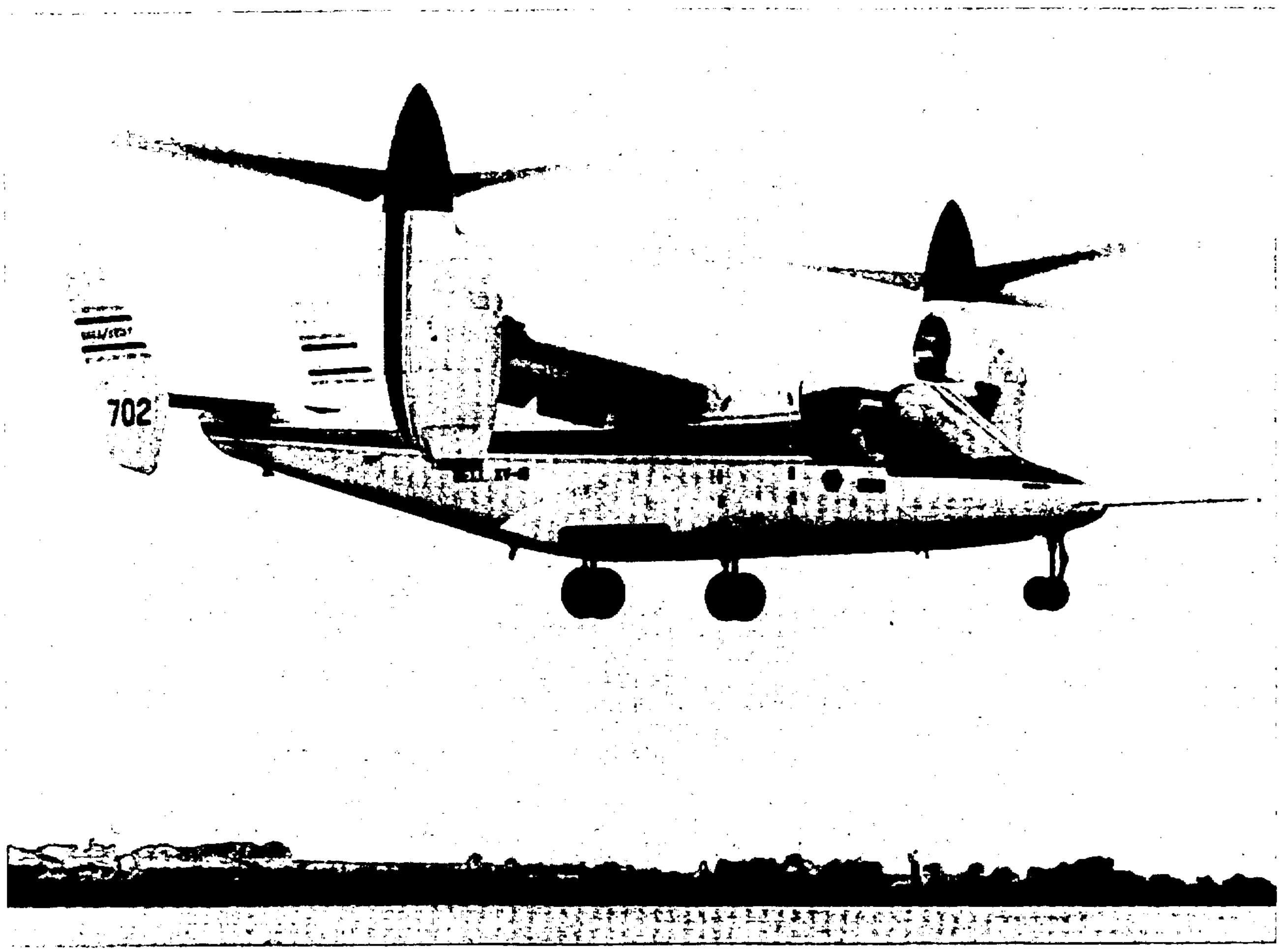
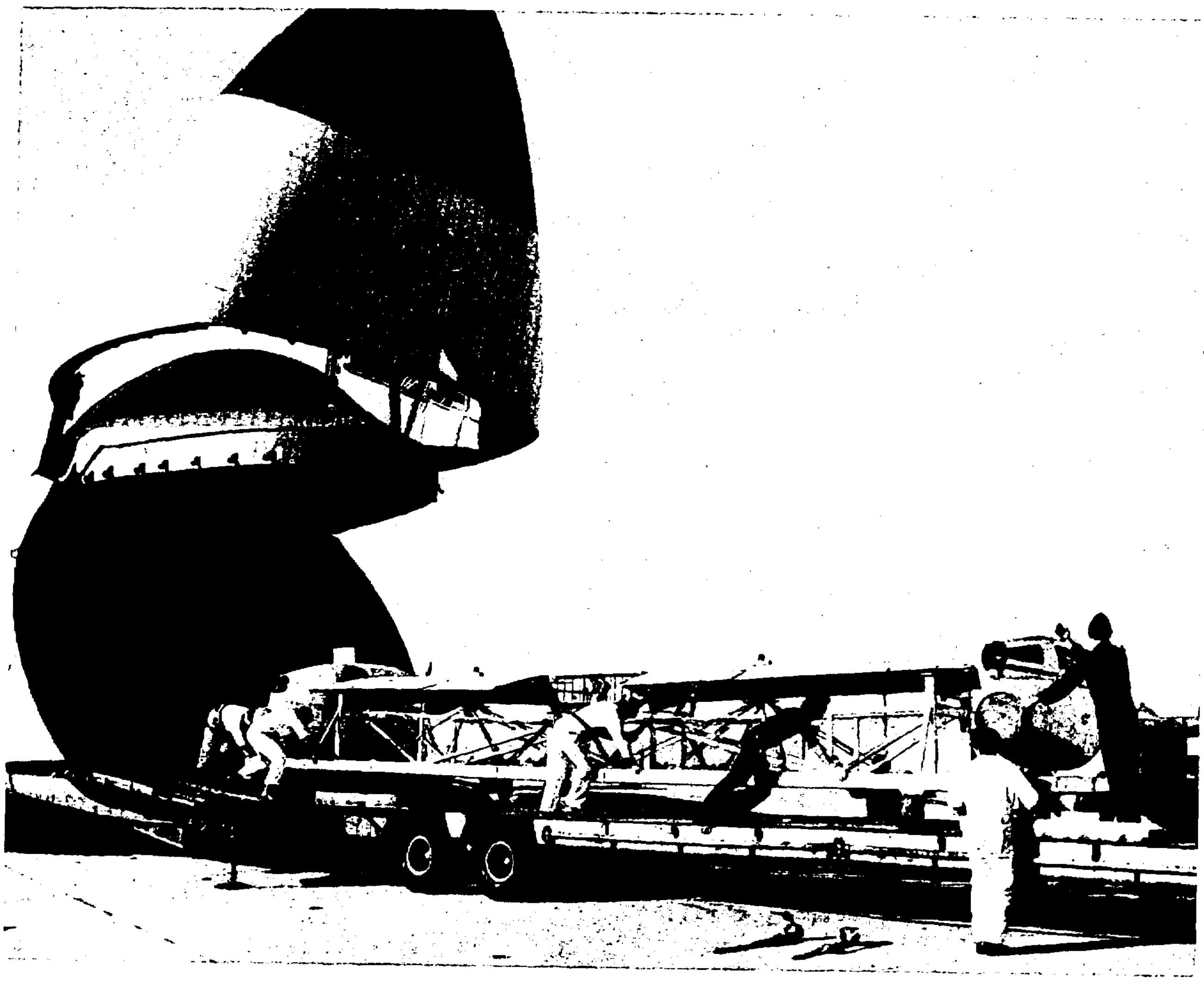


Figure 4.- Aircraft in hover flight.



(a) Wing on shipping cradle.

Figure 5.- Arrival of the aircraft at Ames.

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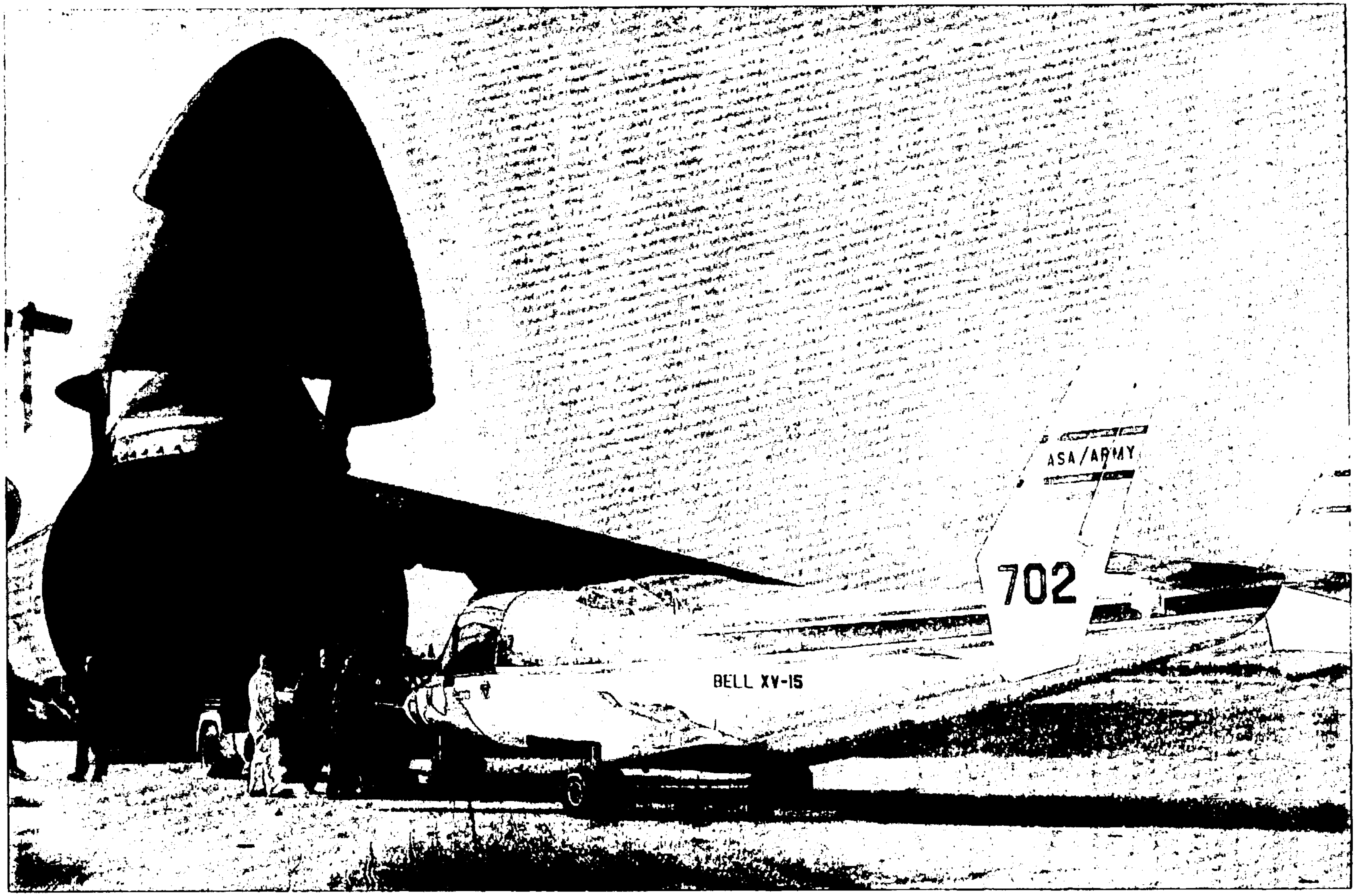

(b) Fuselage.

Figure 5.— Concluded.

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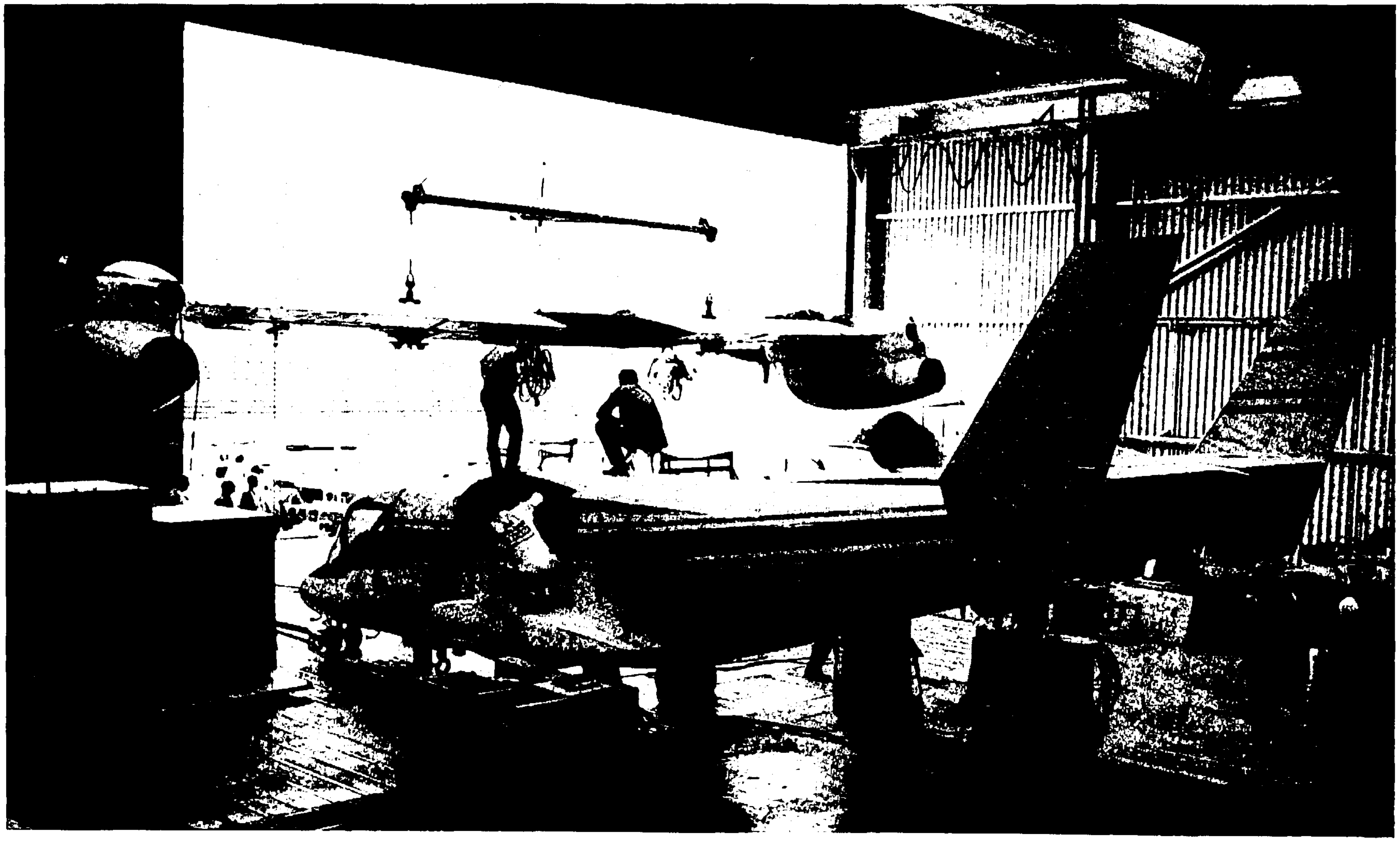

Figure 6.— Assembly of aircraft.

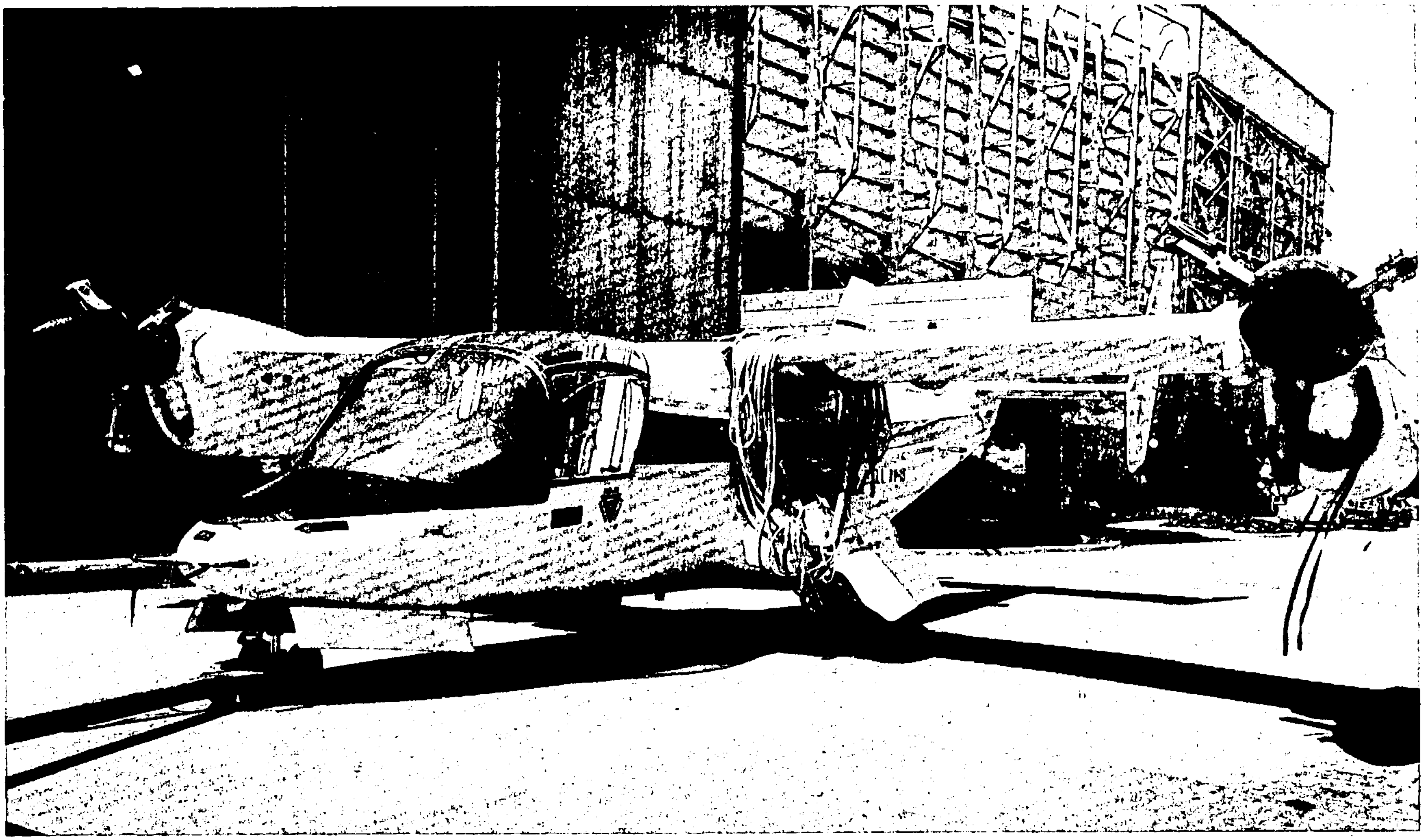

Figure 7.— Moving the aircraft to the tunnel.

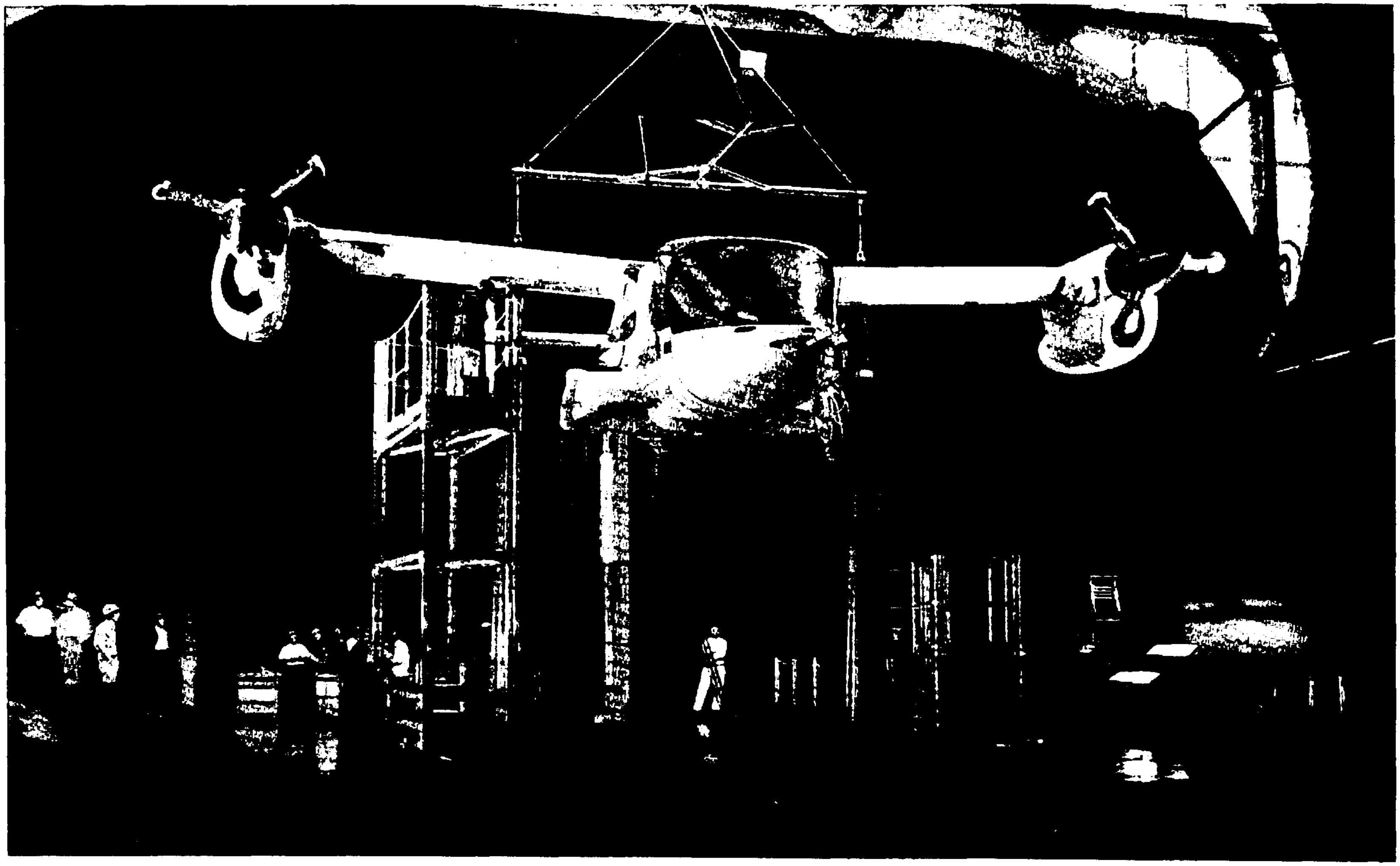

Figure 8.‒ Installing aircraft on the support struts.

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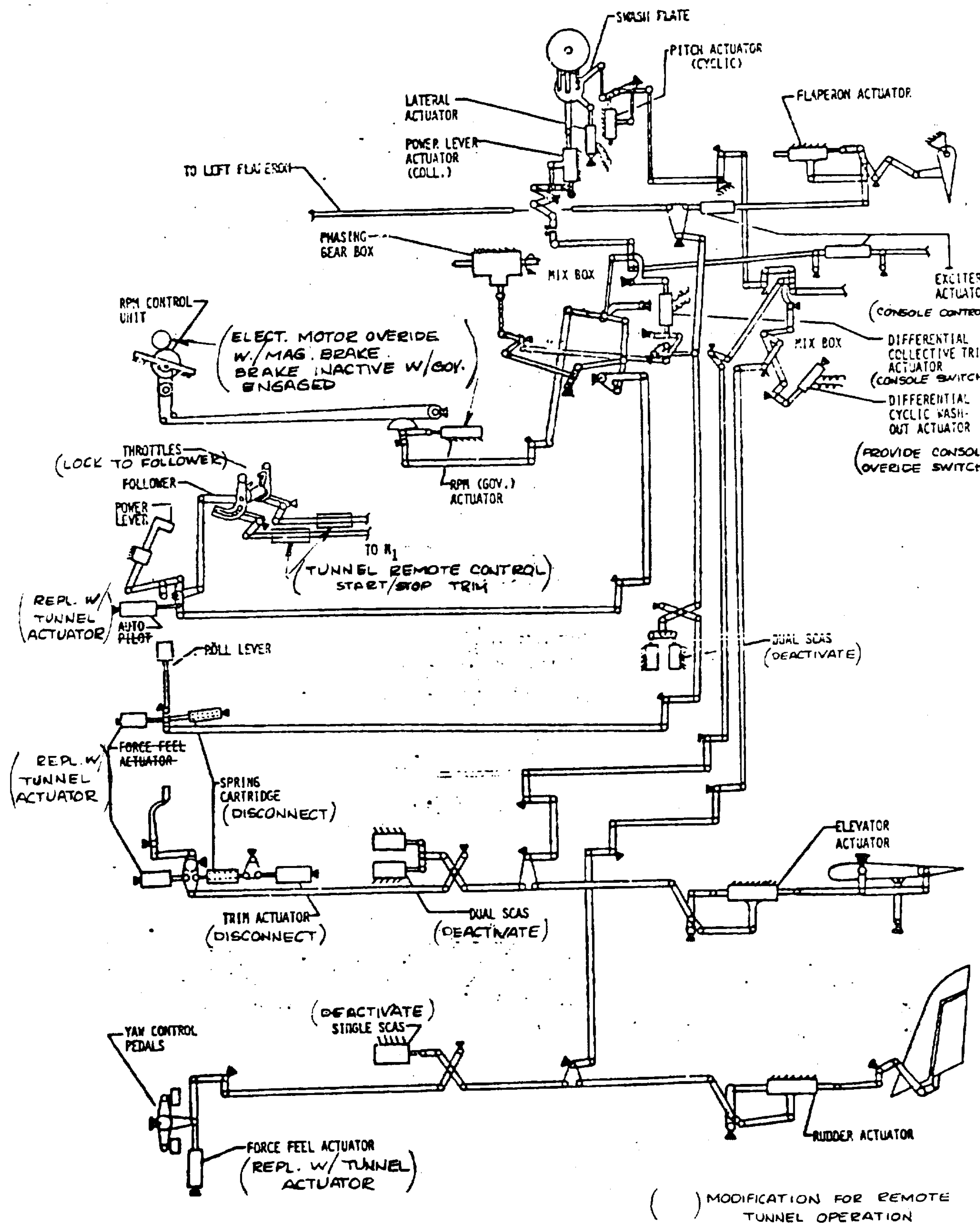


Figure 9.— Control system for remote operation.

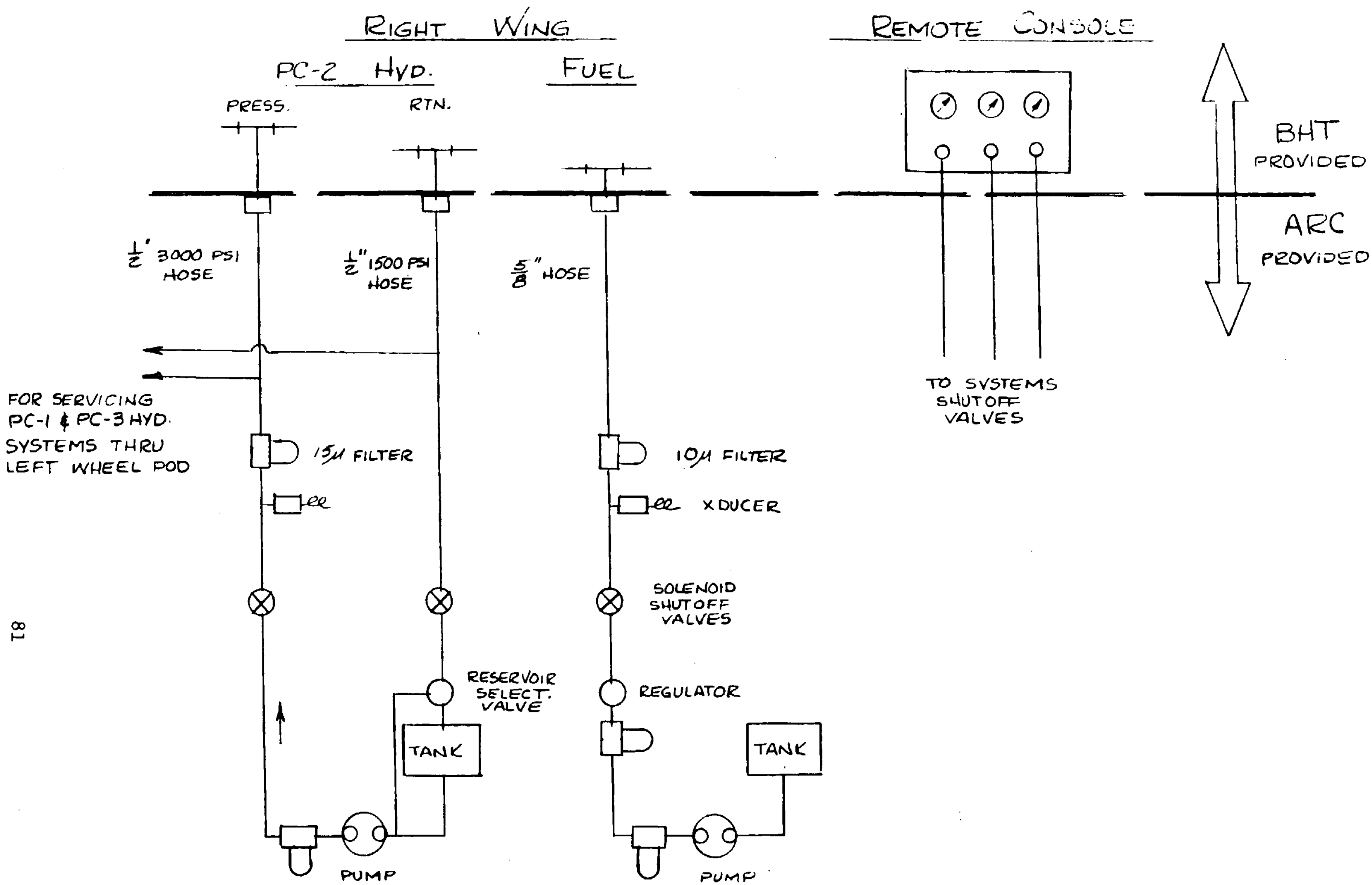


Figure 10.- Aircraft/wind-tunnel interface, hydraulic & fuel systems.

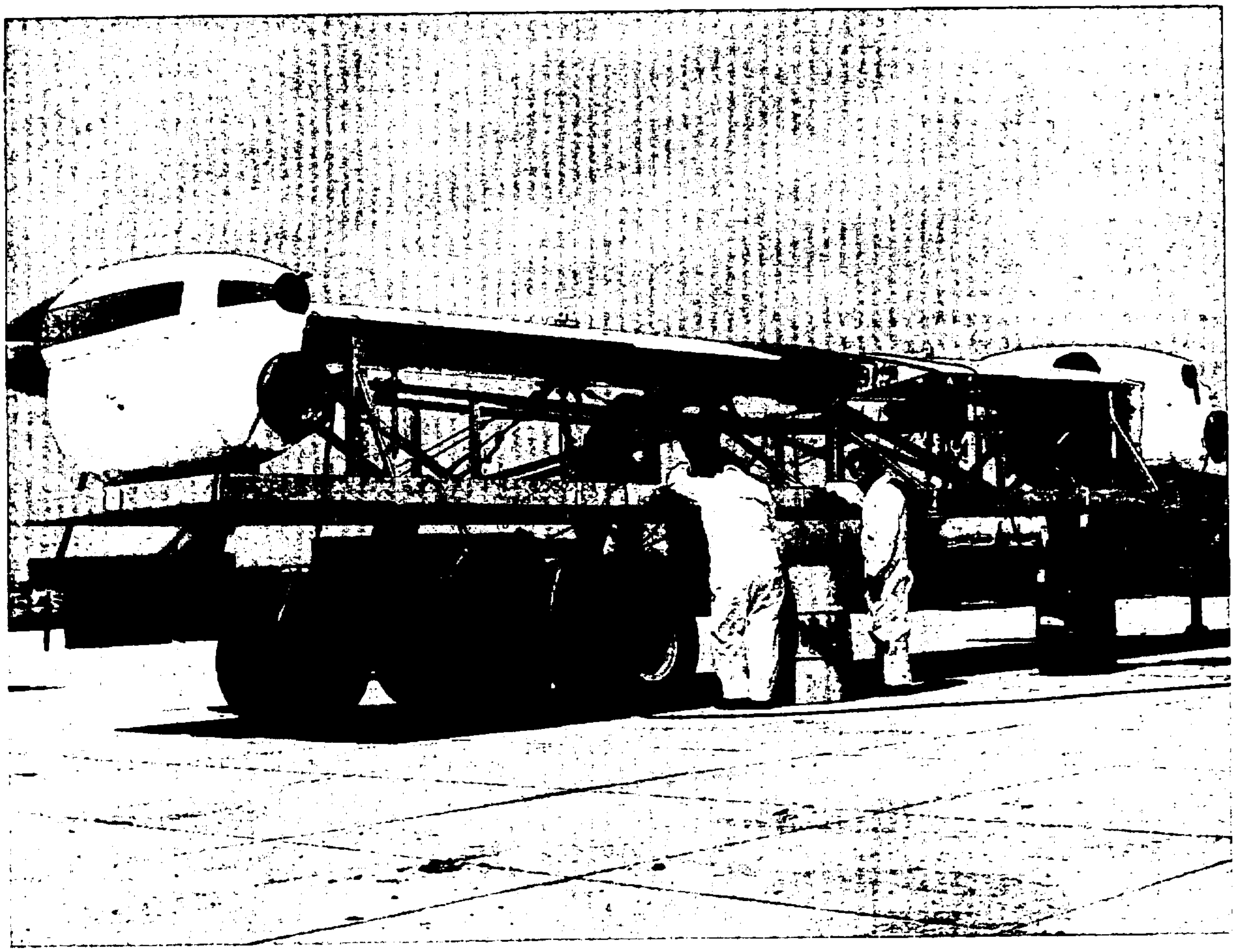
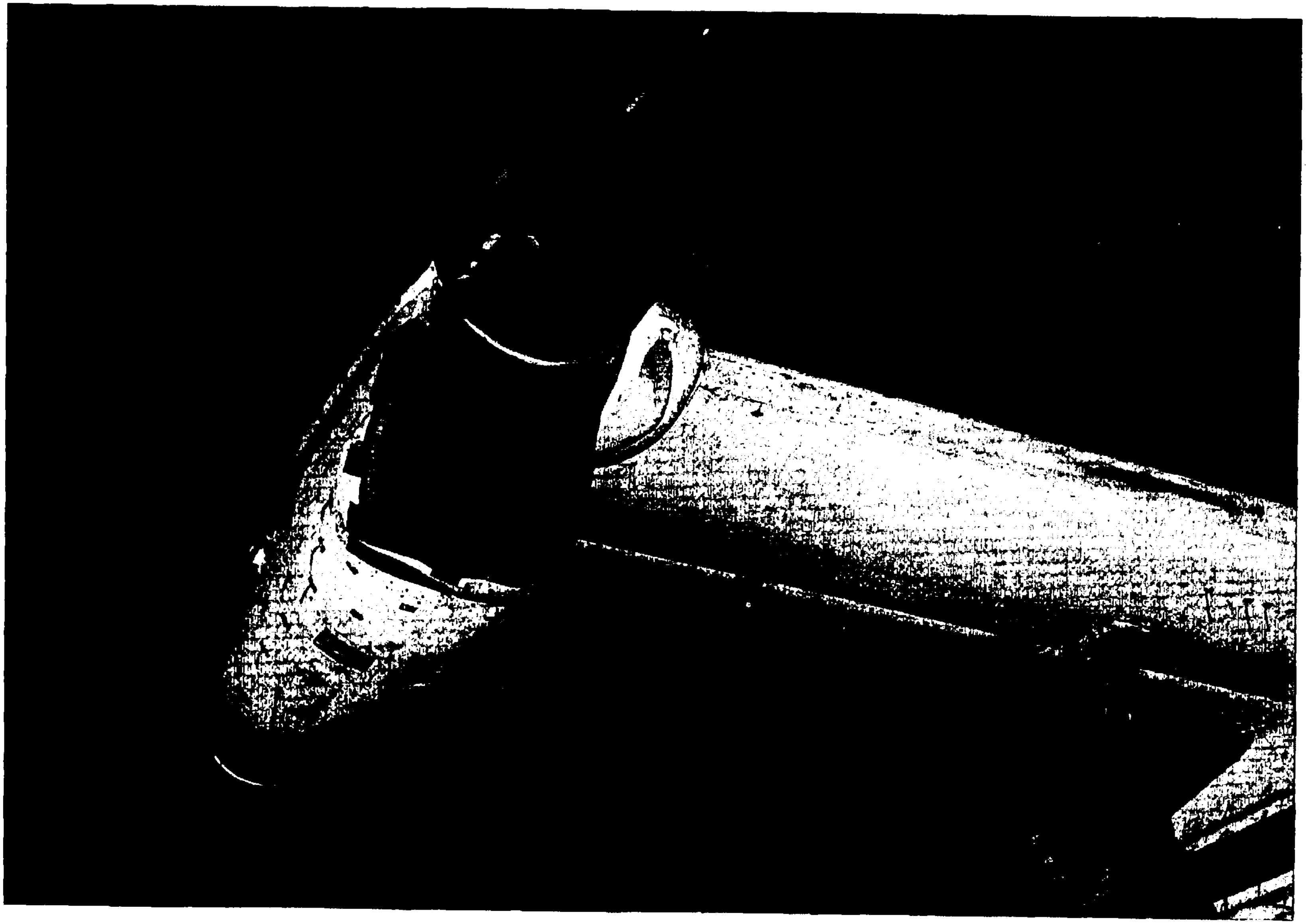


Figure 11.- Purging fuel from tanks.



(a) Inlet.

Figure 12.— Engine airflow seals for power-off operation.

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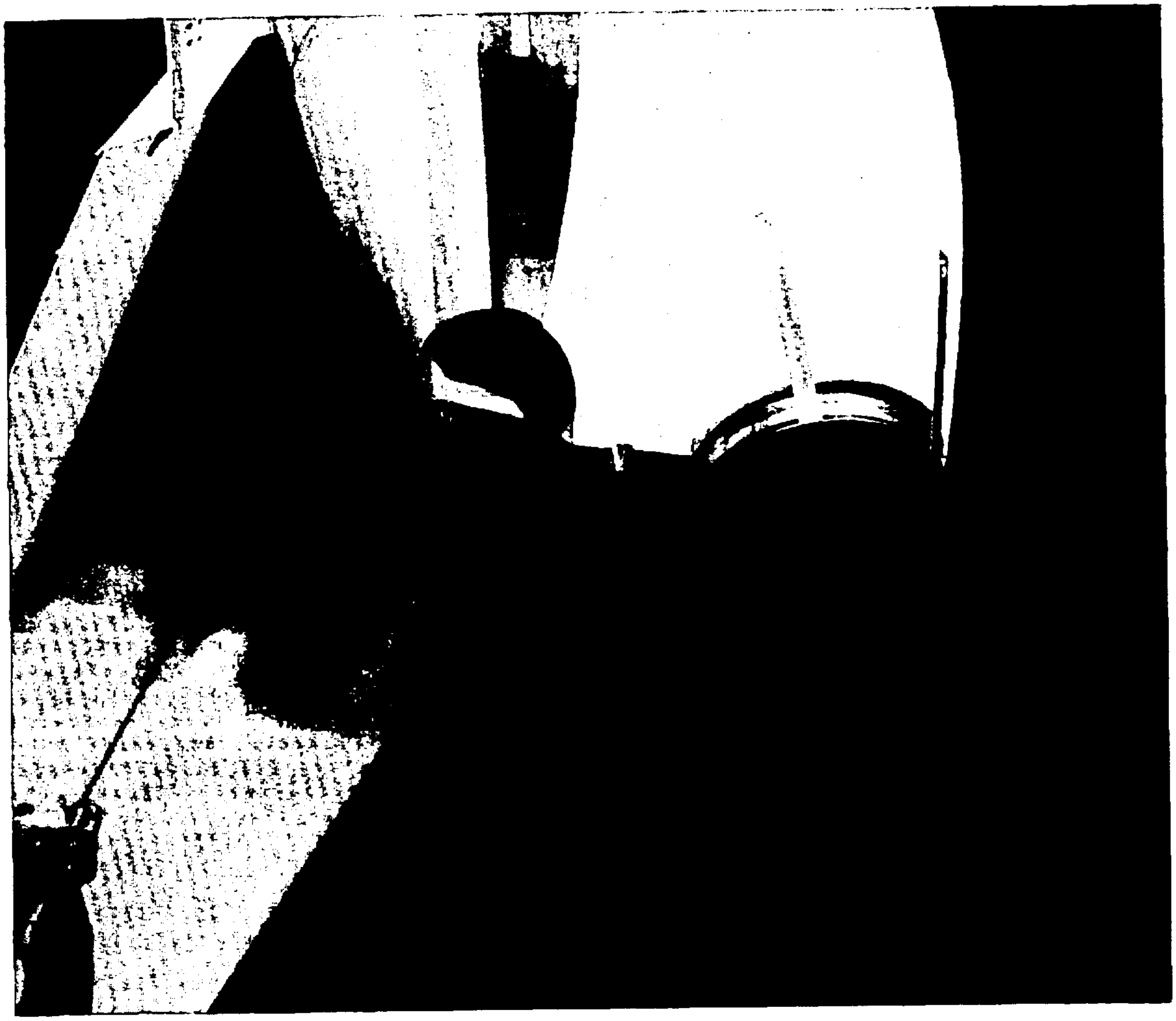
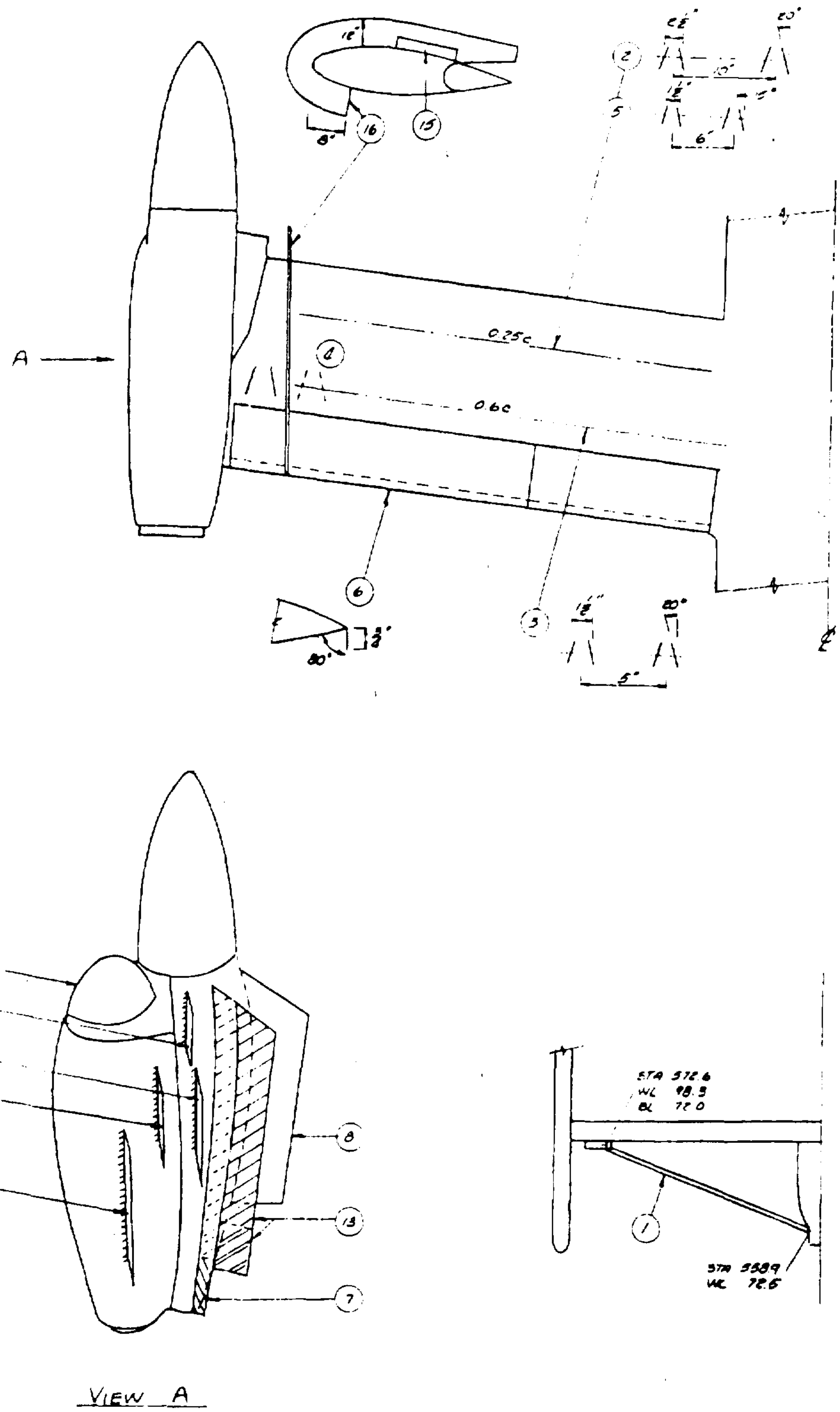


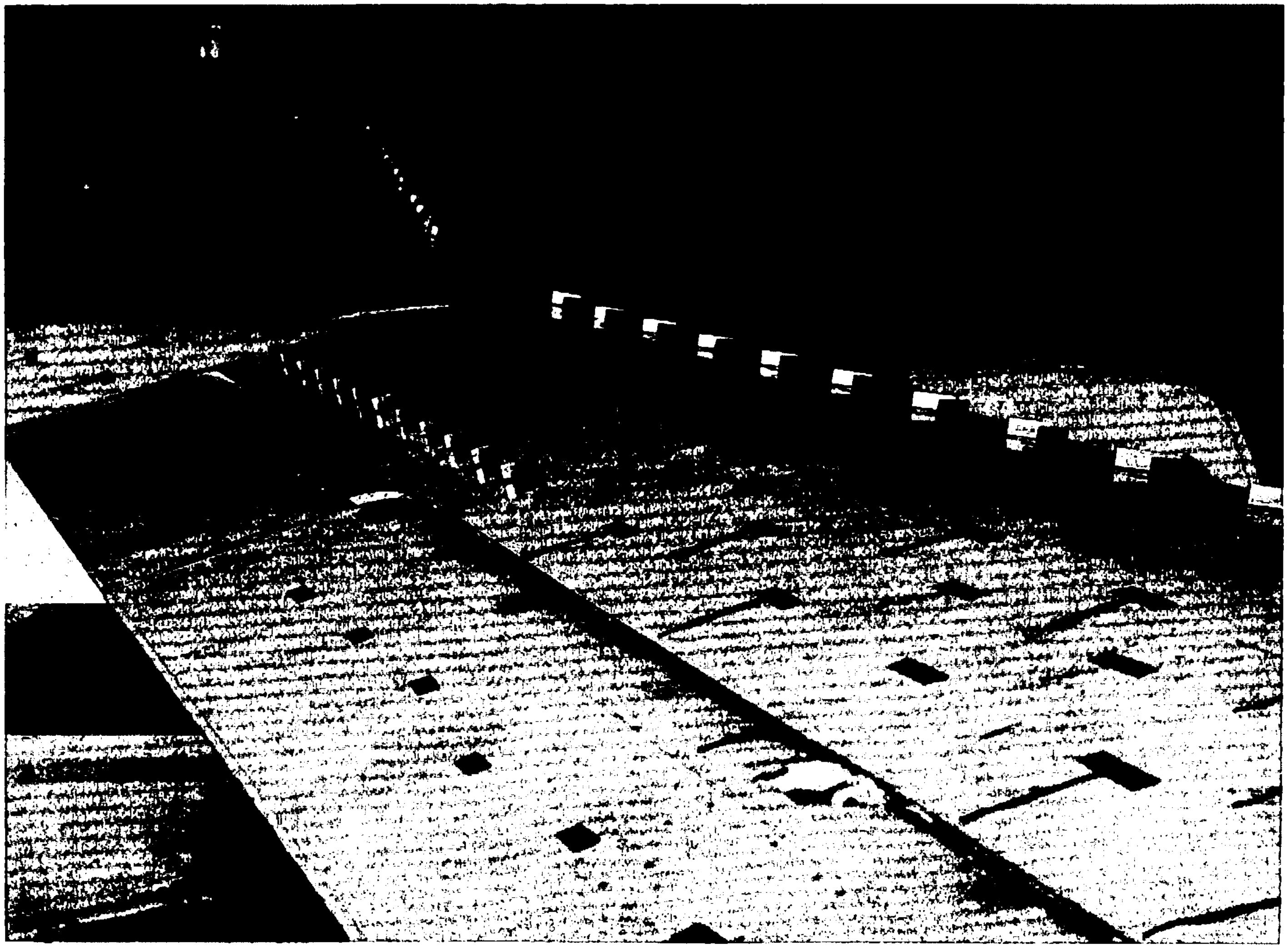
(b) Tail pipe.

Figure 12.- Concluded



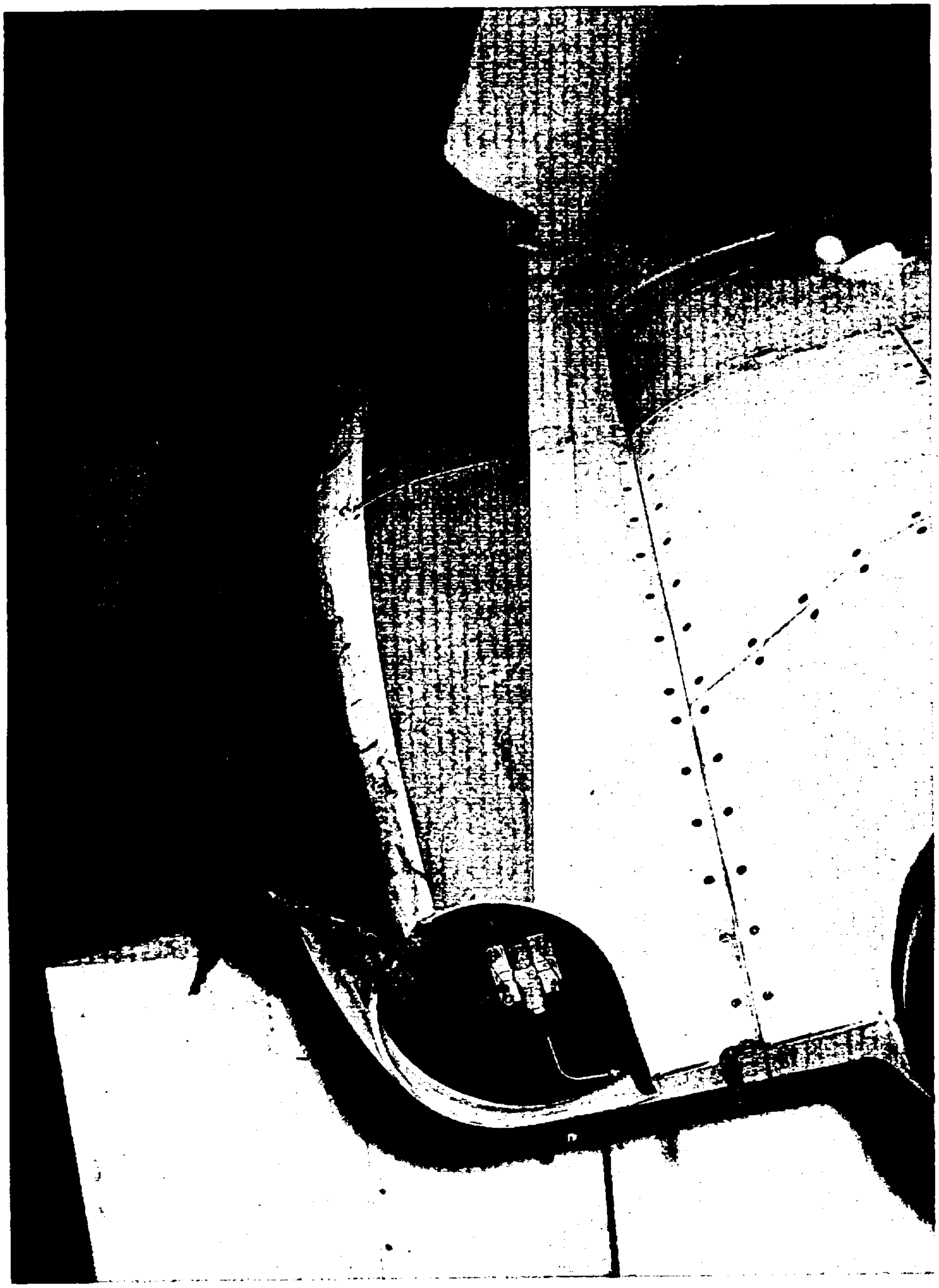
(a) Geometry (ref. Table II).

Figure 13.- Aircraft configuration modifications.



(b) Vortex generators.

Figure 13.- Continued



(c) Fence 7.

Figure 13.— Continued.

(d) Horizontal tail struts.

Figure 13.- Concluded.

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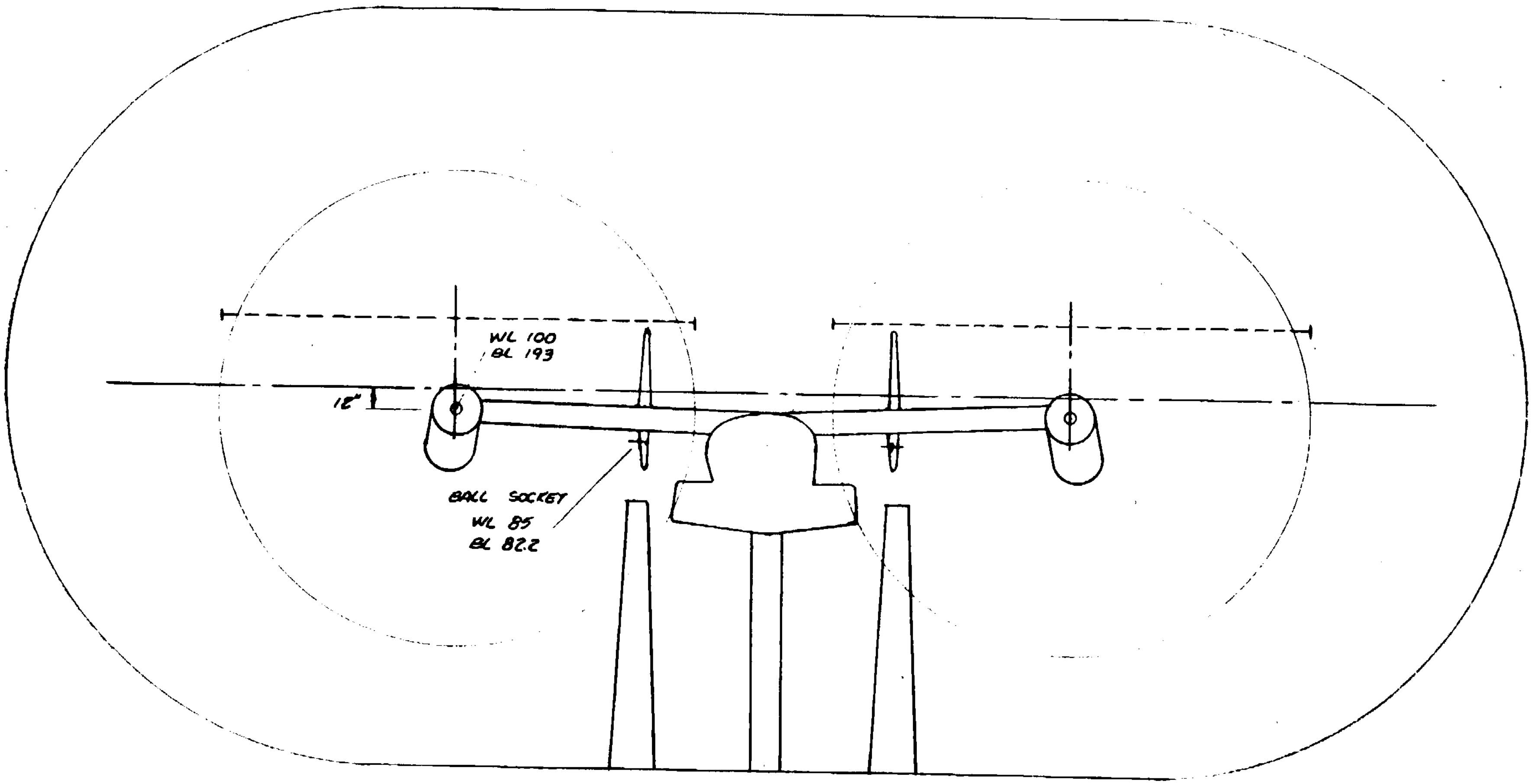
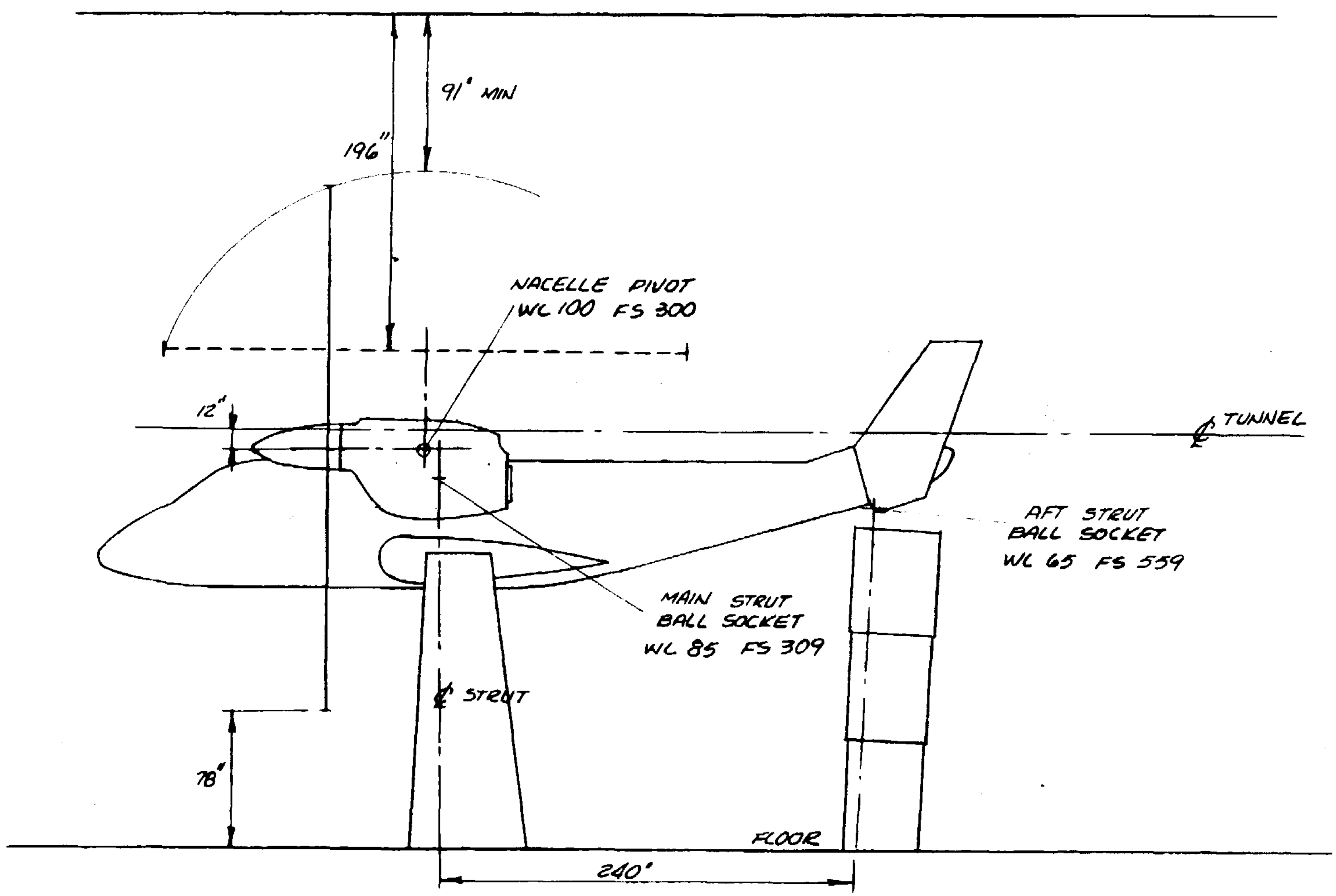


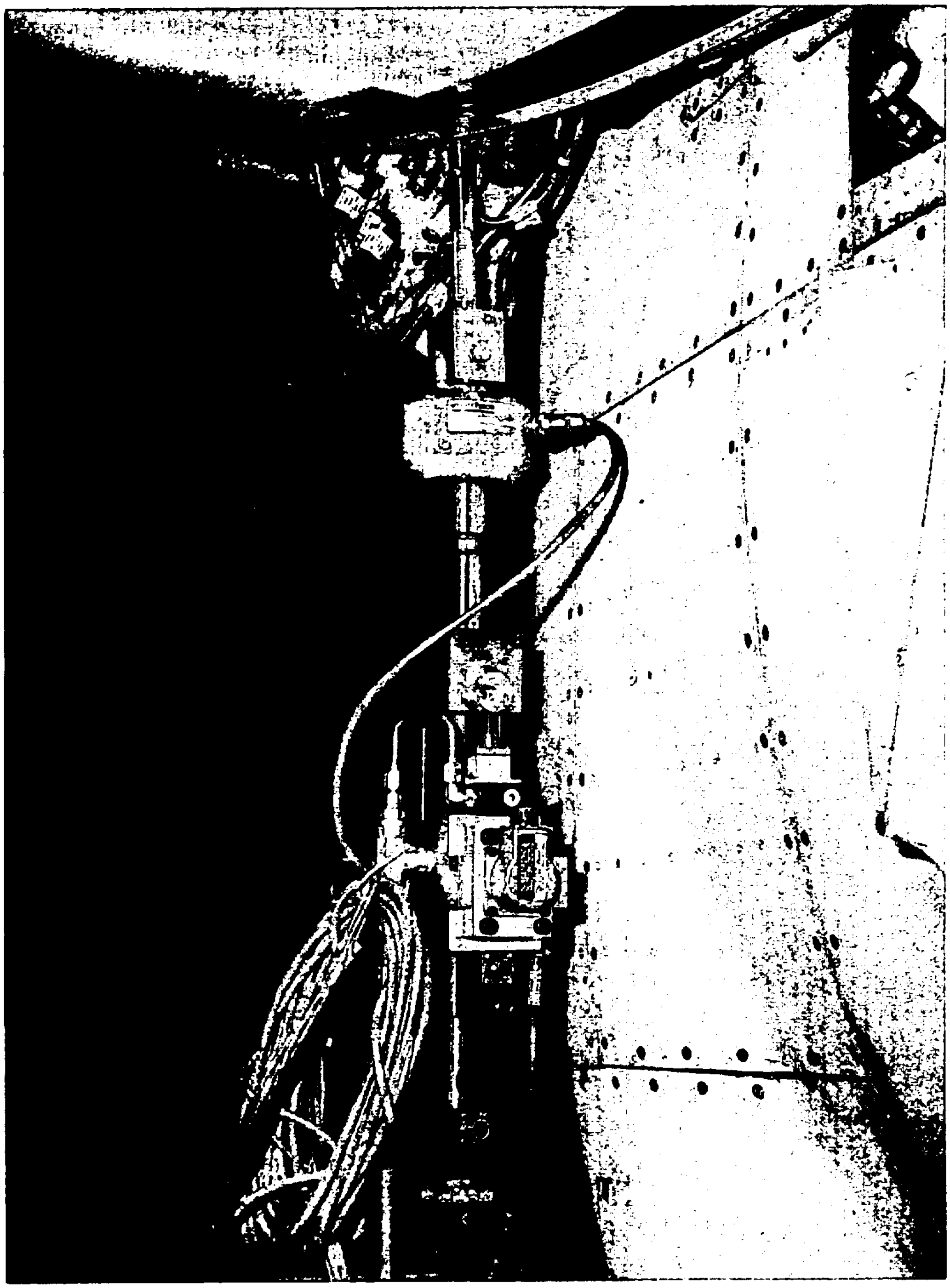
(a) Front view.

Figure 14.— The XV-15 in the 40- by 80-foot wind tunnel.



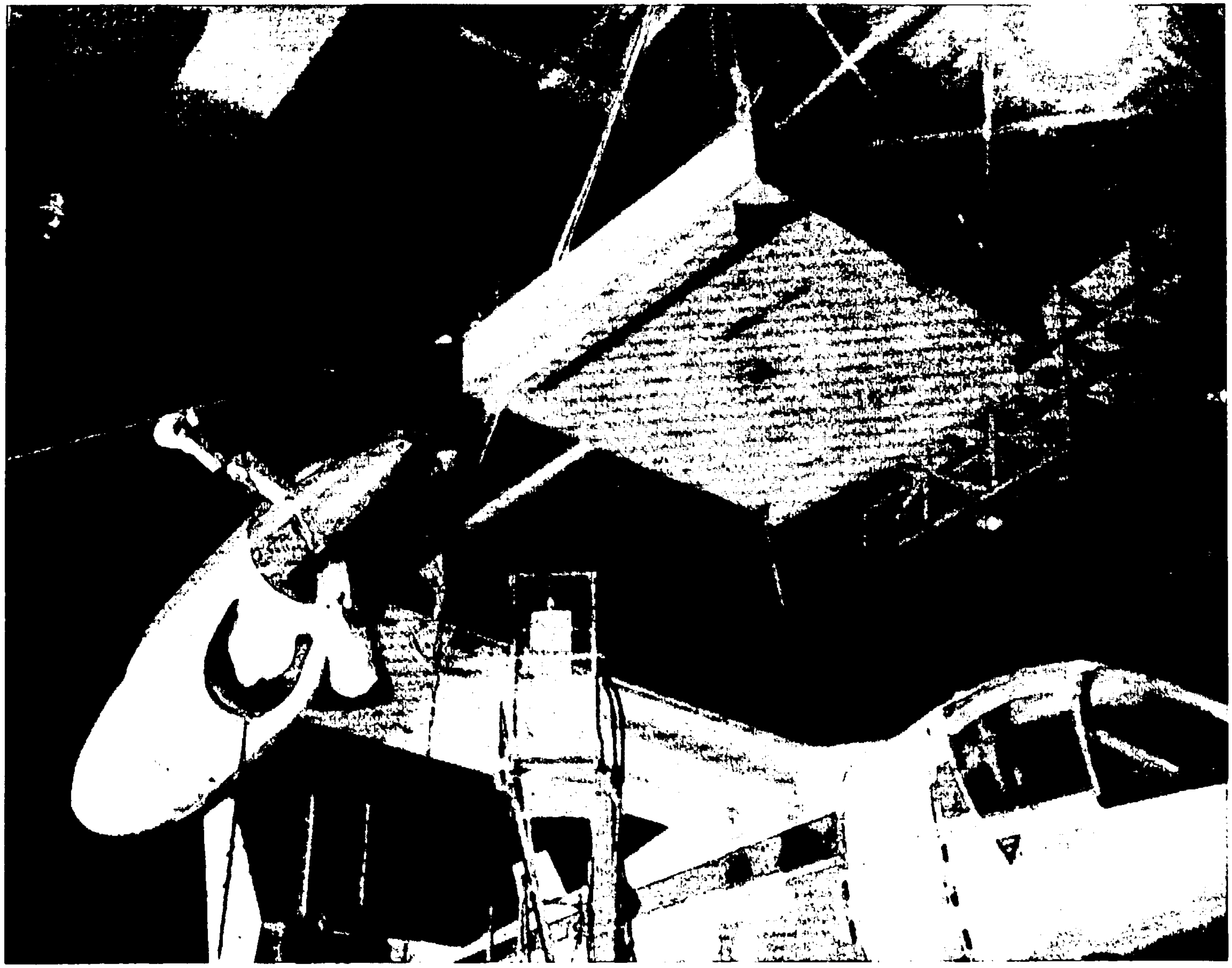
(b) Side view.

Figure 14.— Concluded.



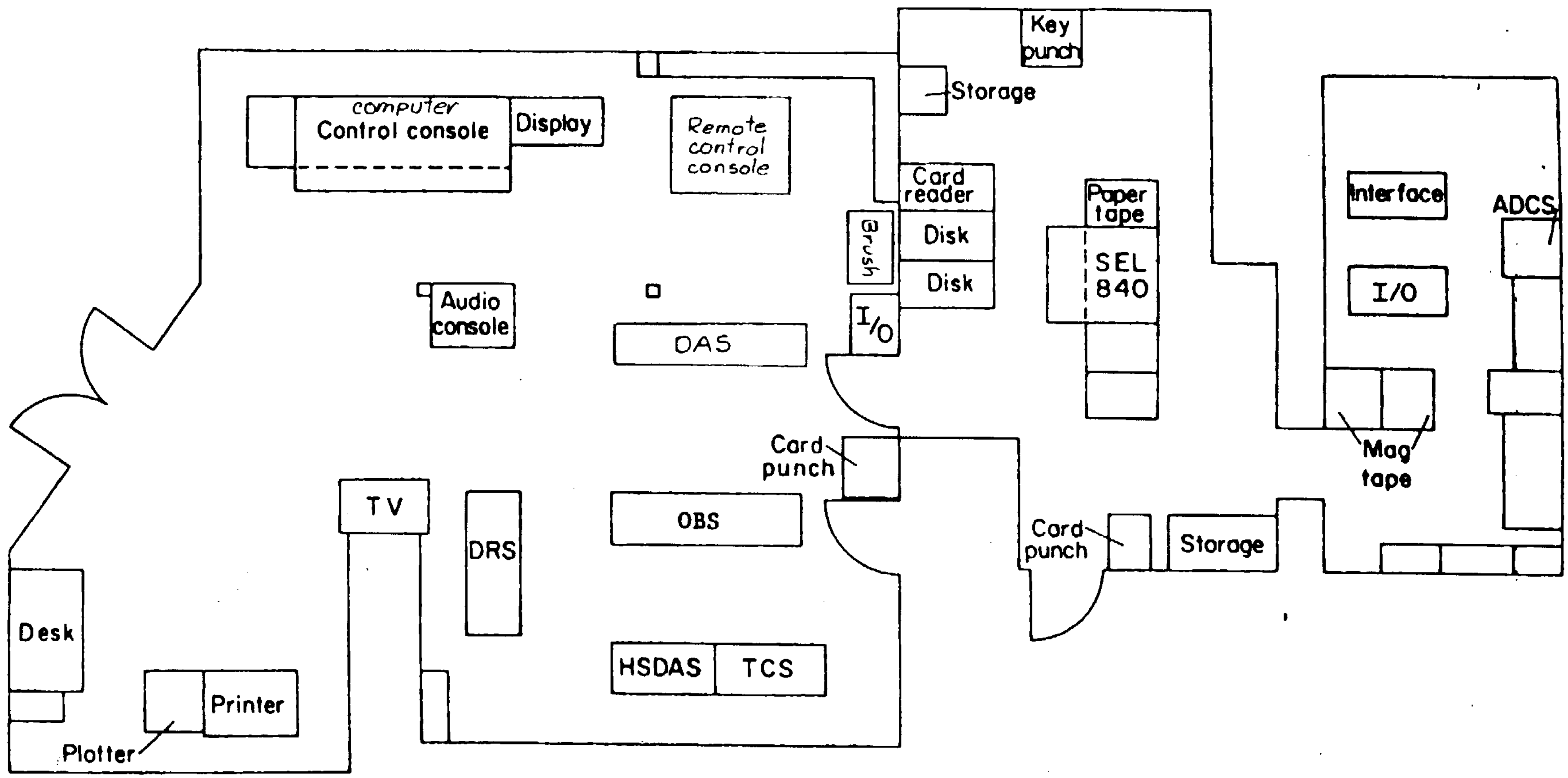
(a) Vertical.

Figure 15.— Shaker installation.



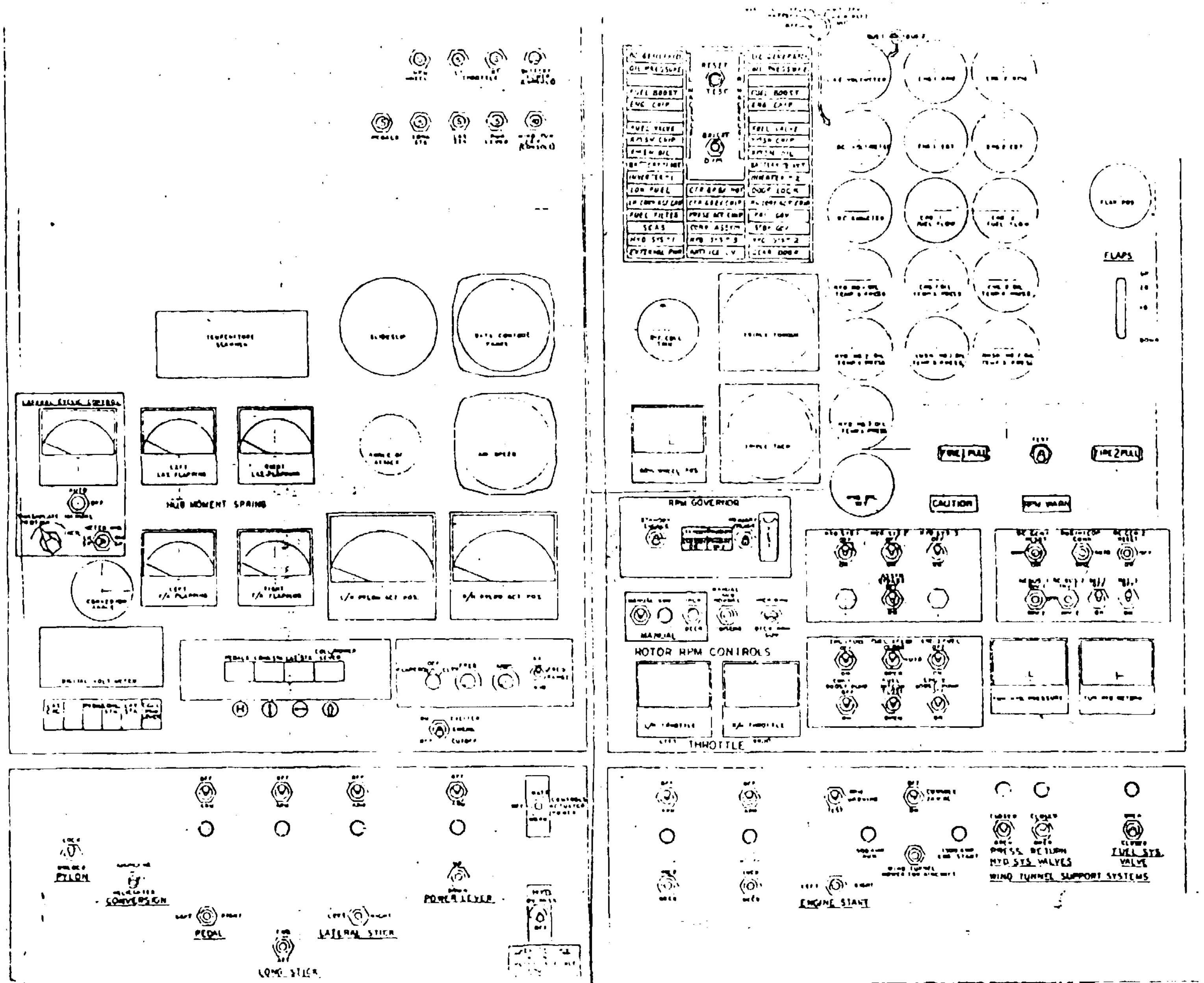
(b) Horizontal.

Figure 15.- Concluded.



(a) Layout of control room.

Figure 16.— Remote operation of aircraft in the tunnel.



(b) Remote control console.

Figure 16.– Continued.

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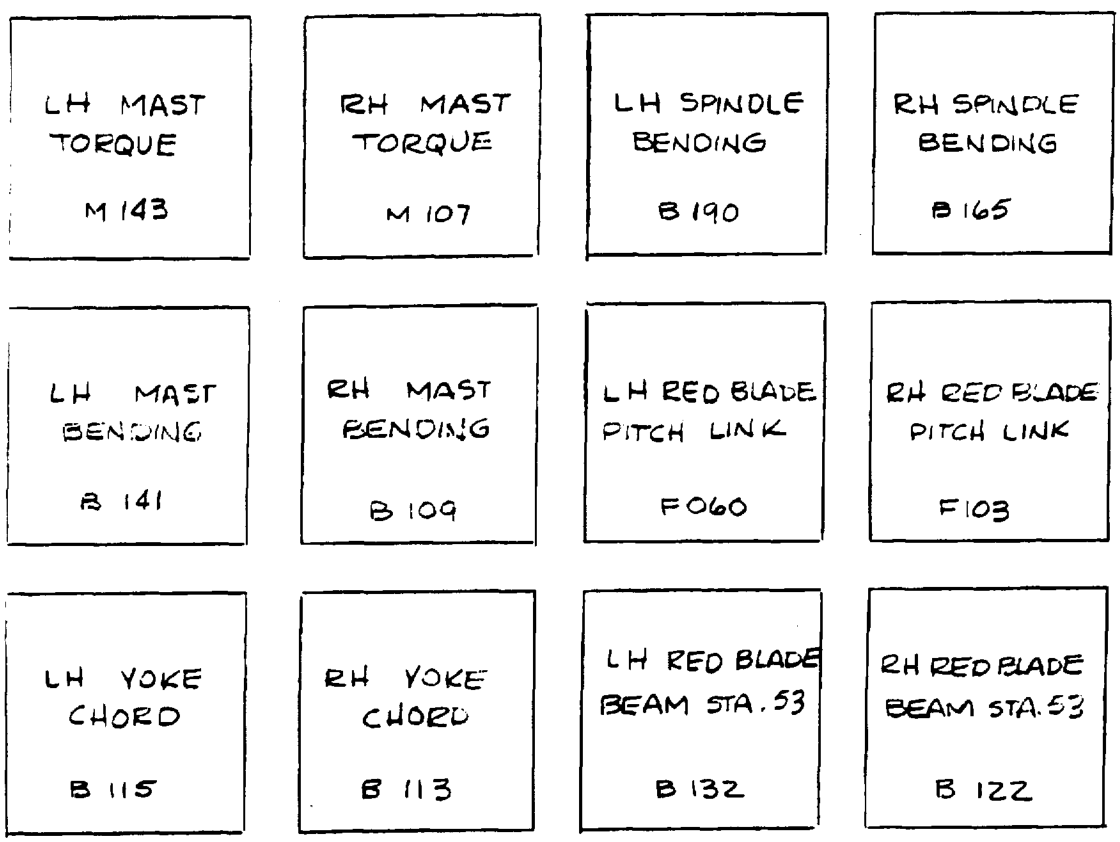
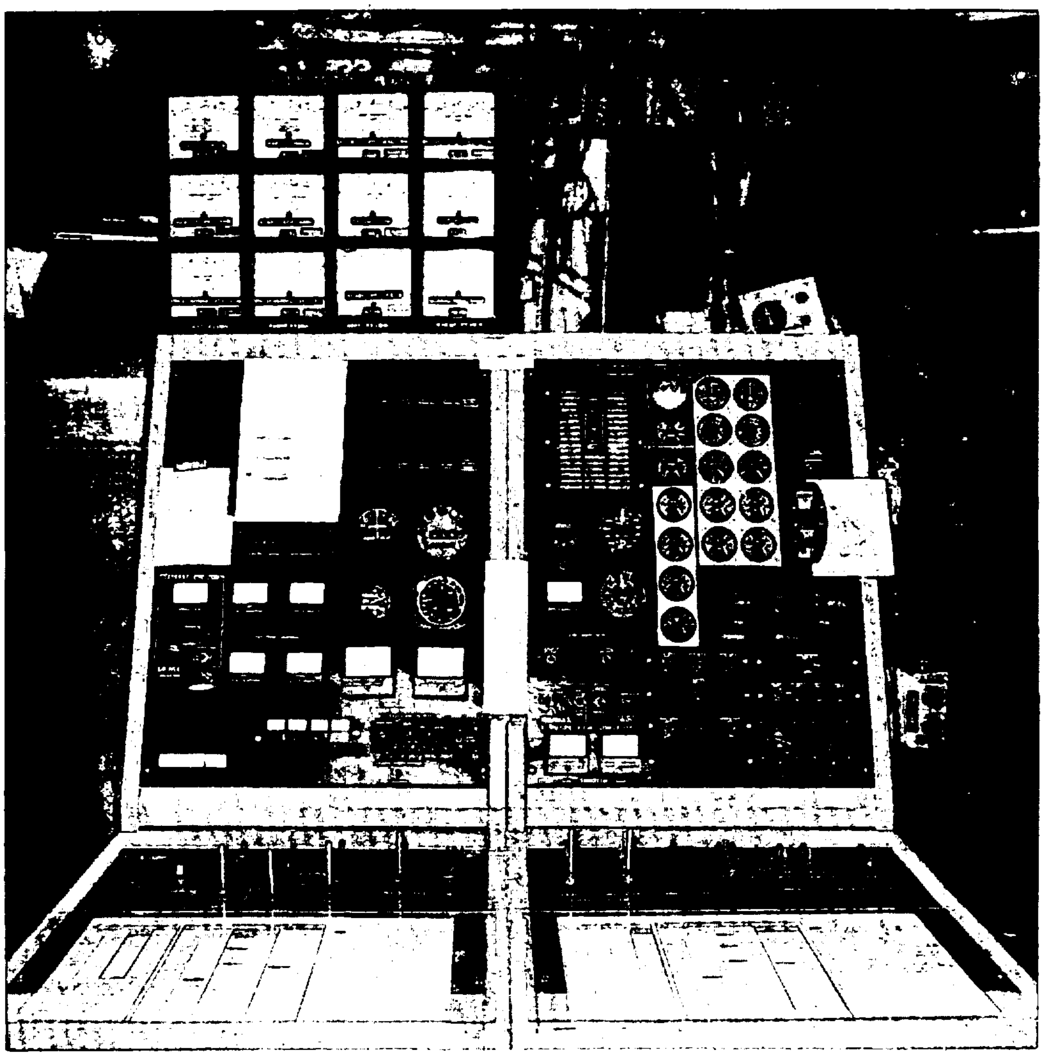


(c) Loads panel on console.

Figure 16.— Continued.



(d) Remote control console.

Figure 16.- Continued.

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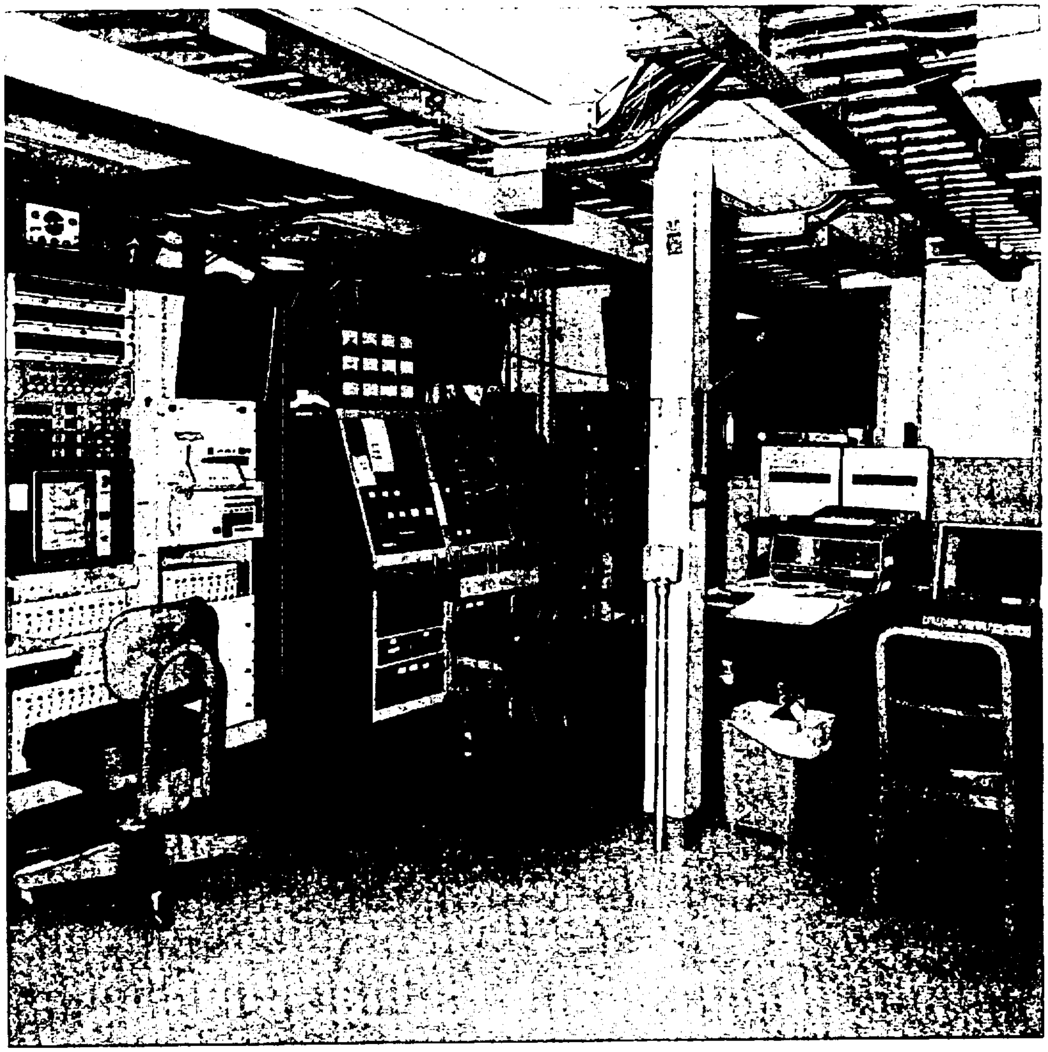

(e) Control room.

Figure 16.- Continued.

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(f) Control room during test operations.

Figure 16.— Concluded.

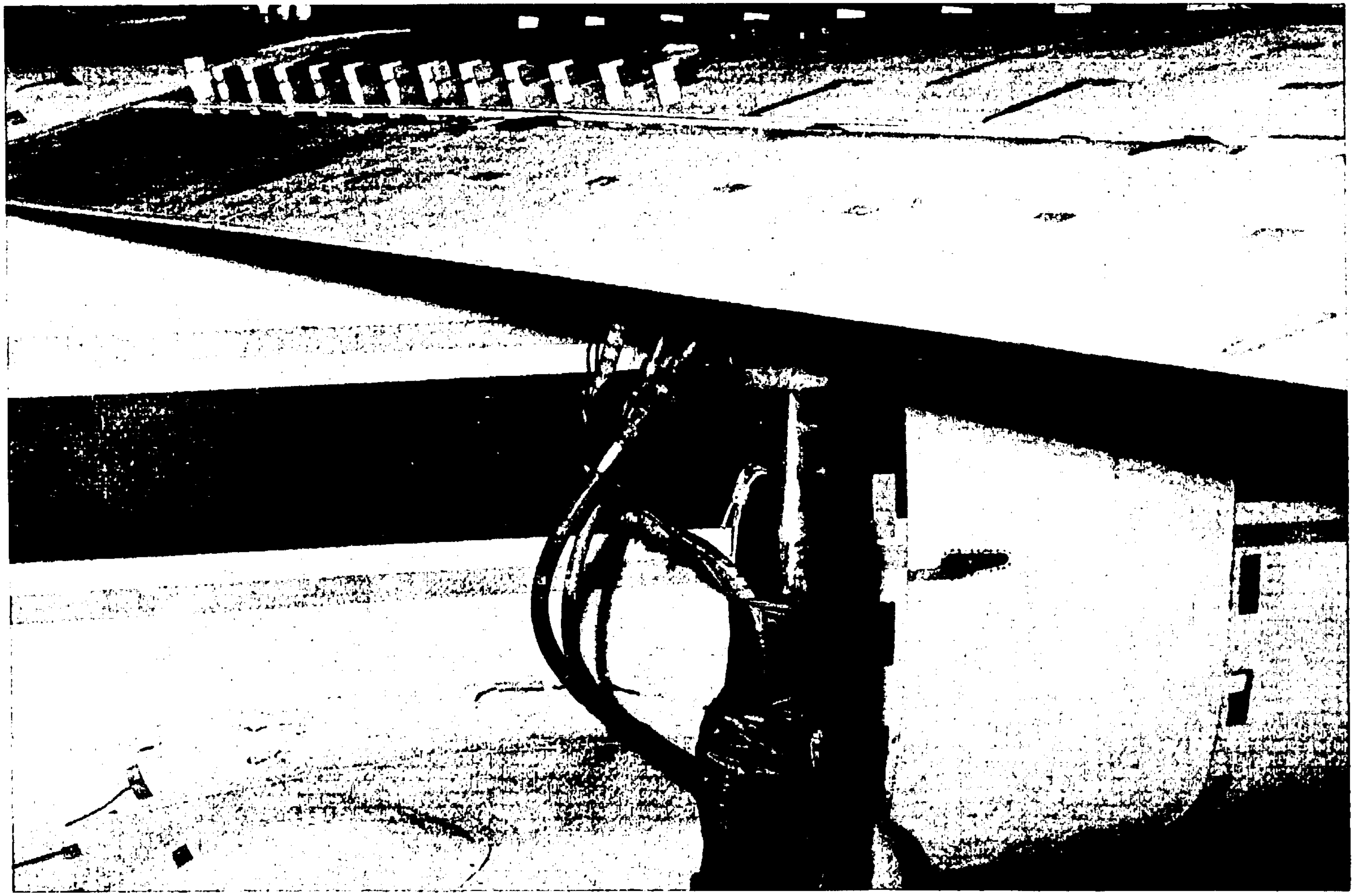

Figure 17.— Right-hand support strut.

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Figure 18.— Left-hand support strut.

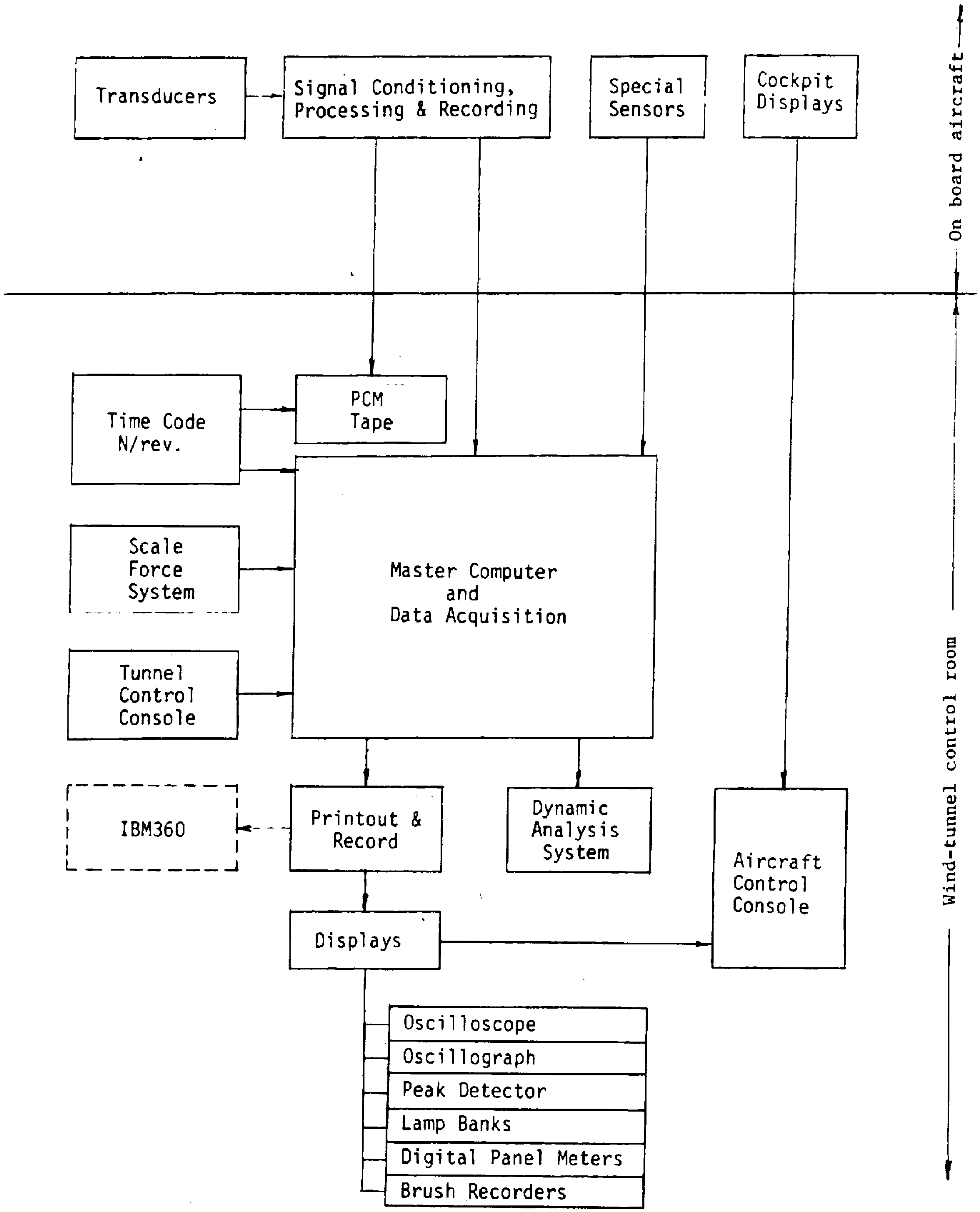

Figure 19.— Data system block diagram.

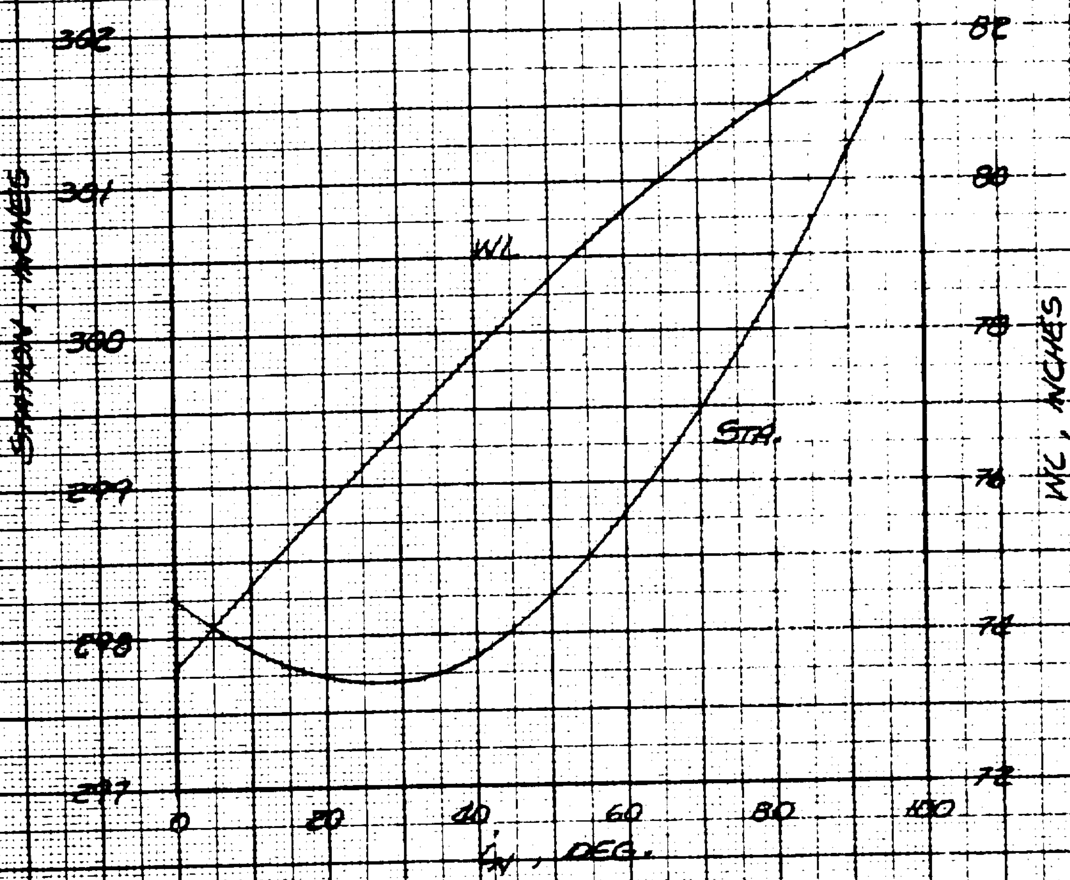

FIGURE 20.- MOMENT CENTER VARIATION WITH NACELLE TILT.

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RUN 20 SEQ 19 06-JUN 09:46:20

ALPH	-8.00	MTL	6.209E+04
PSI	0.000	MTR	6.258E+04
I N	90.0		
		HPL	556.
VKTS	79.7	HPR	560.
RPM	564.		
VOR	0.182	N/6	-4.696E+03
TEMP	88.0	R/6	304
		Y/6	-43.7
COLL	23.1		
LAT	50.6	L/6	1.408E+04
LONG	69.6	D/Q	6.49
PED	40.6	M/6	-4.028E+03

XV-15 TEST 525

Figure 21. Data display on CRT in control room.

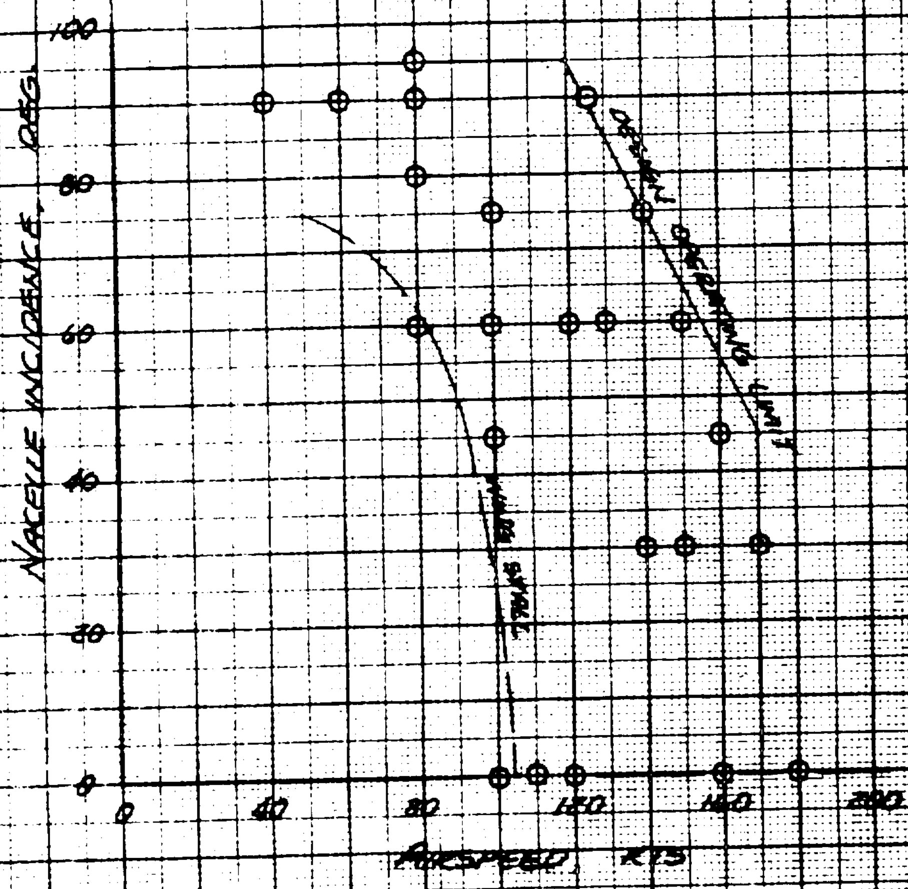
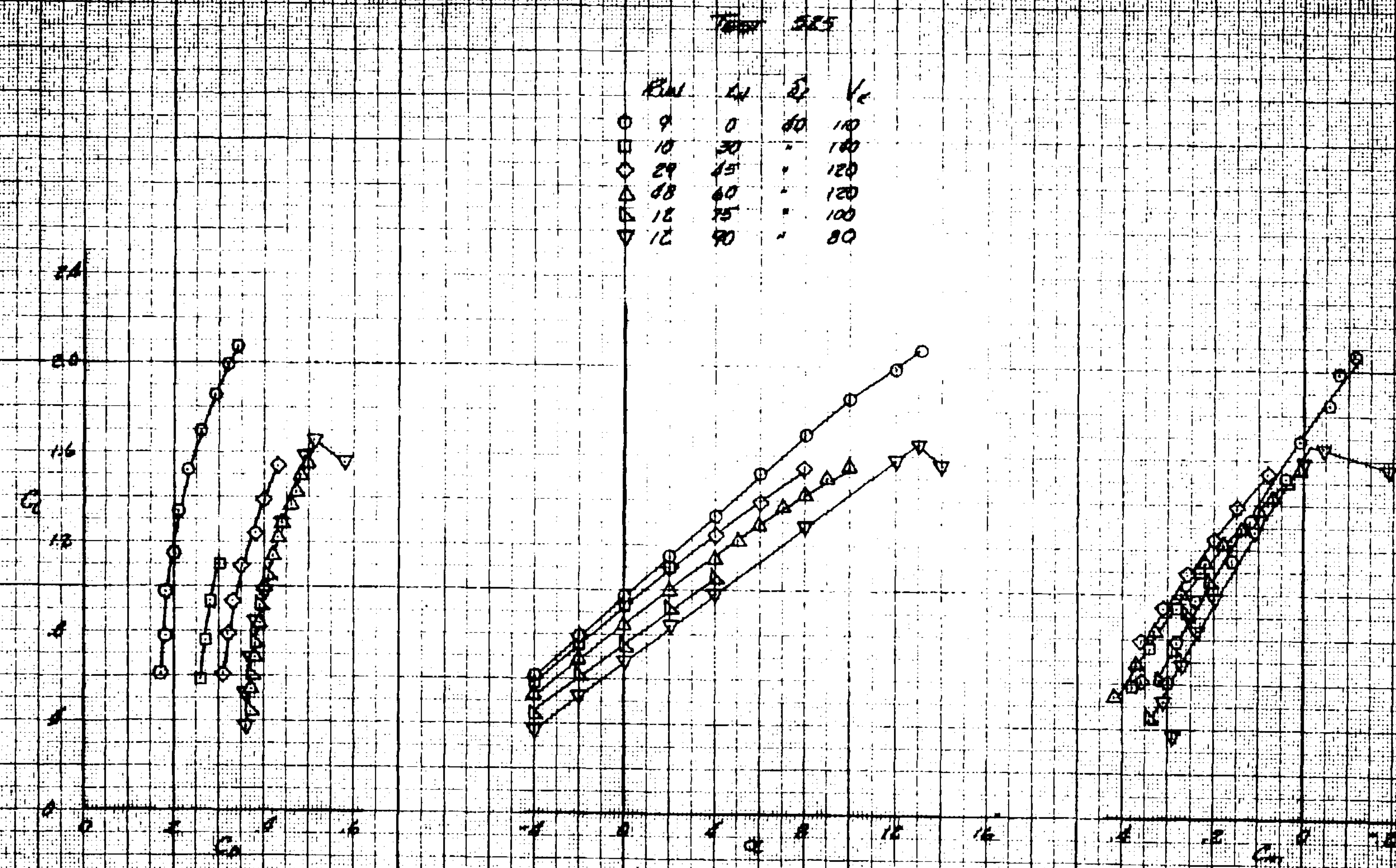
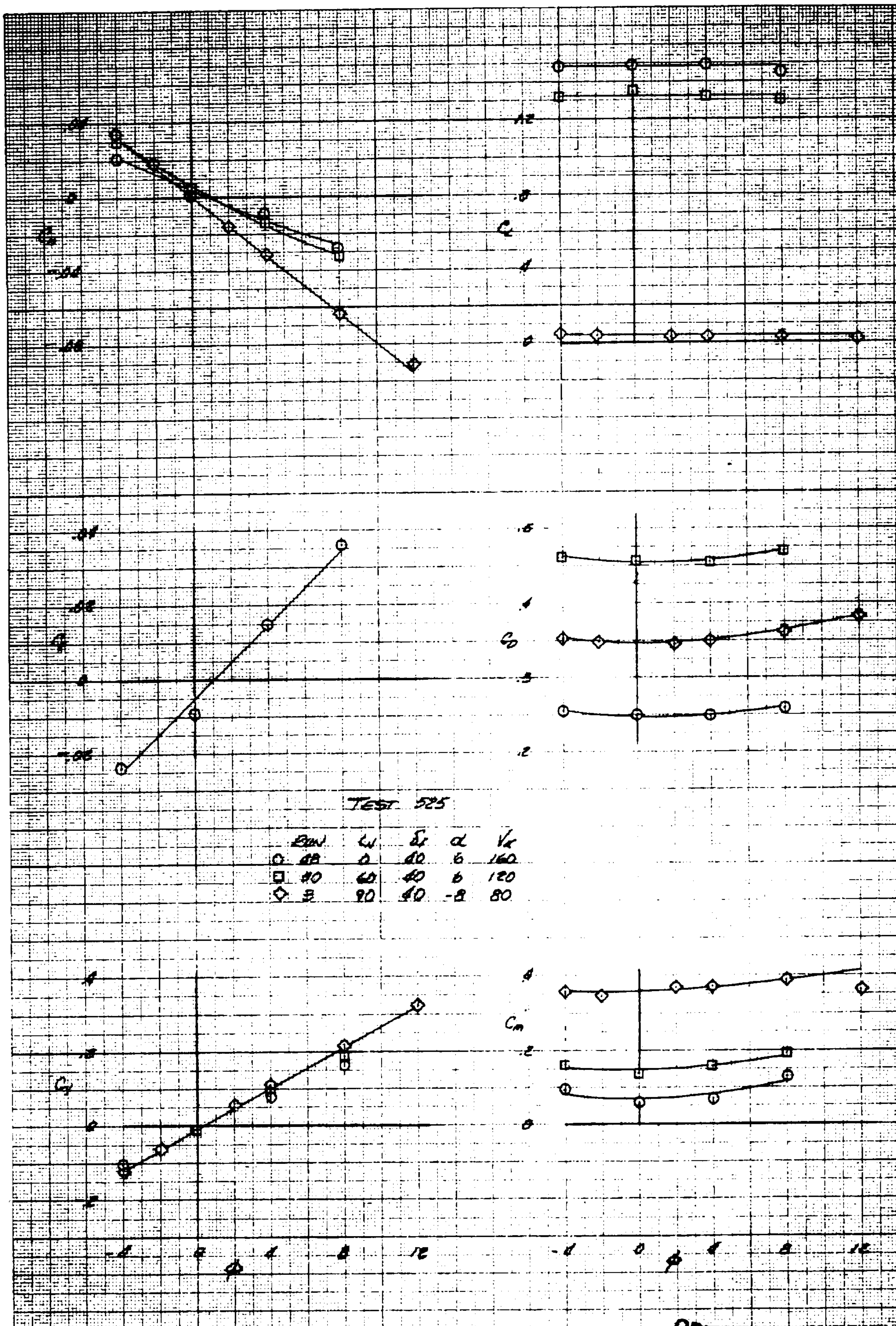


FIGURE 22. - CONVERSION CORRIDOR TEST POINTS.



(a) Longitudinal characteristics

Figure 23.- Effect of nacelle incidence, props off.



(b) CHARACTERISTICS IN YAW.

FIGURE 23 - CONCLUDED.

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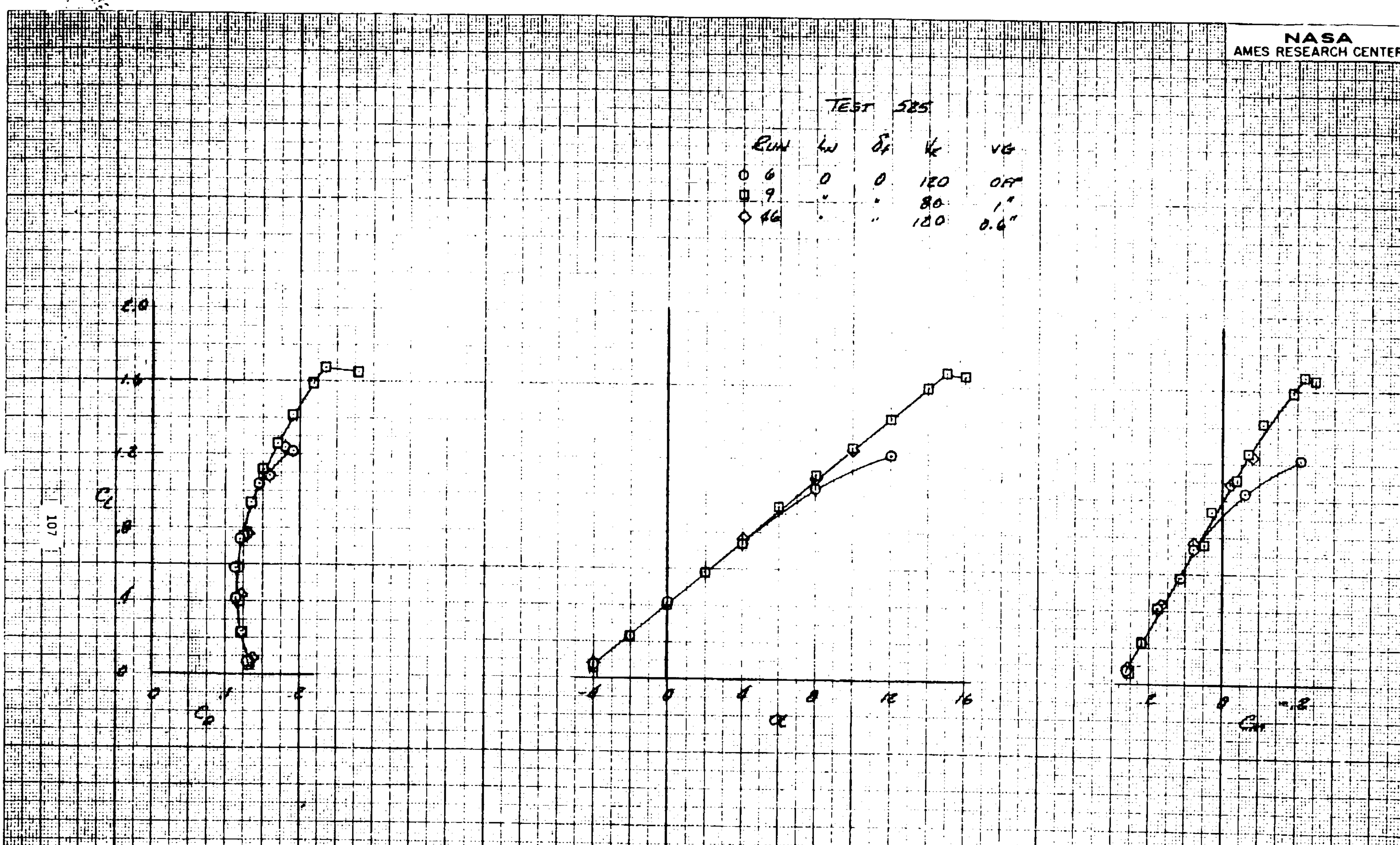

(a) At 0.25c

FIGURE 24.- EFFECT OF VORTEX GENERATORS

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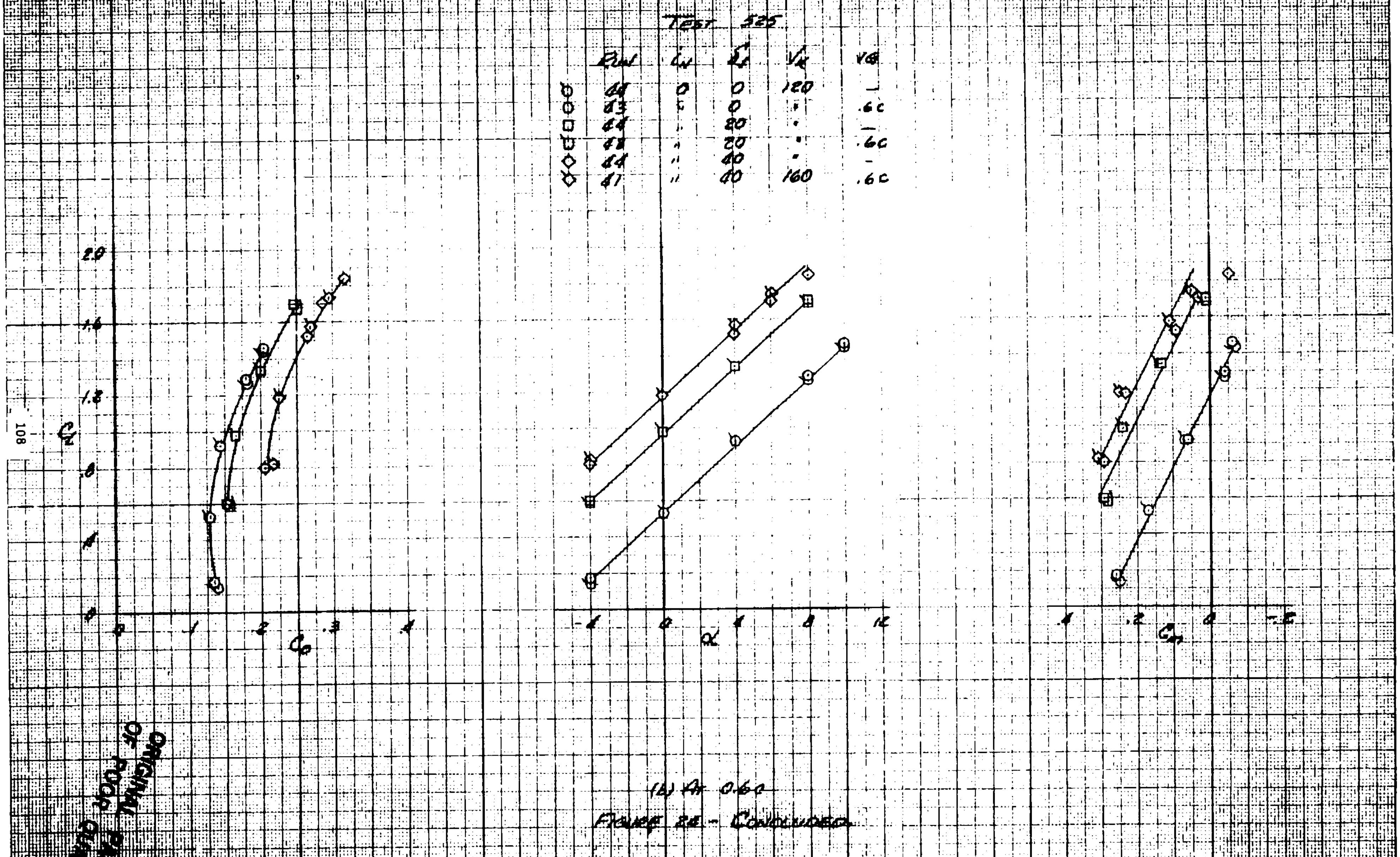

(b) At 0.6c

Figure 22 - Concluded

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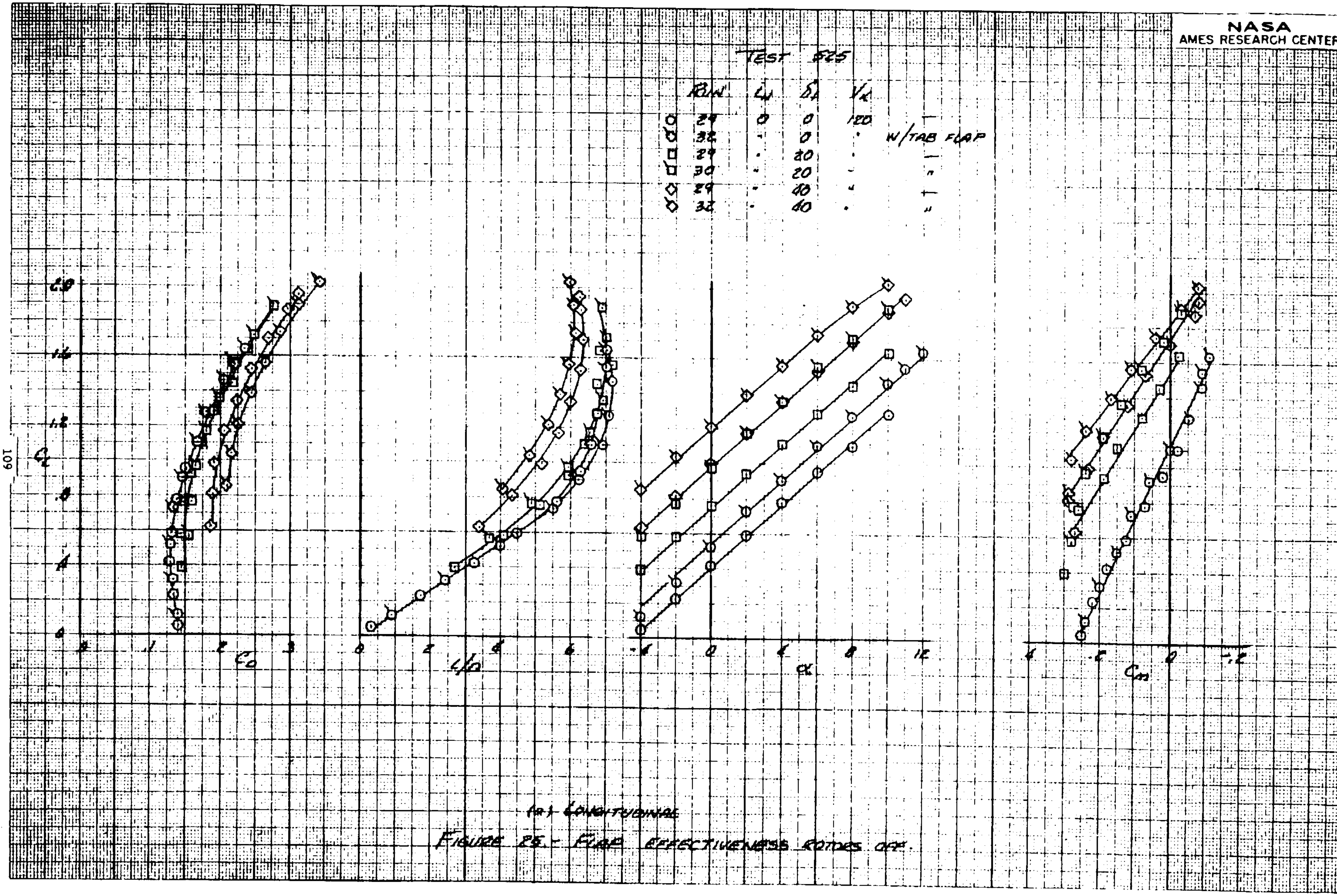
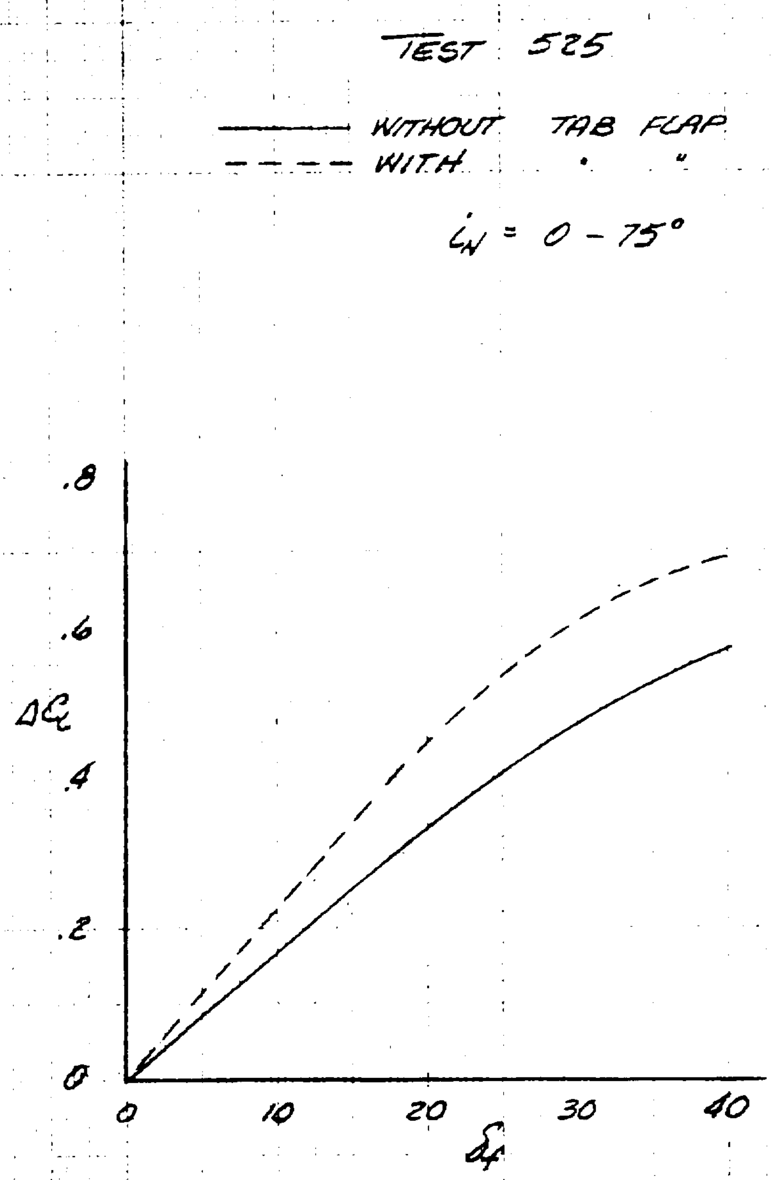


(a) Longitudinal

Figure 25.- Flap effectiveness rotors off.



(b) FLAP LIFT INCREMENT

FIGURE 25.- CONCLUDED.

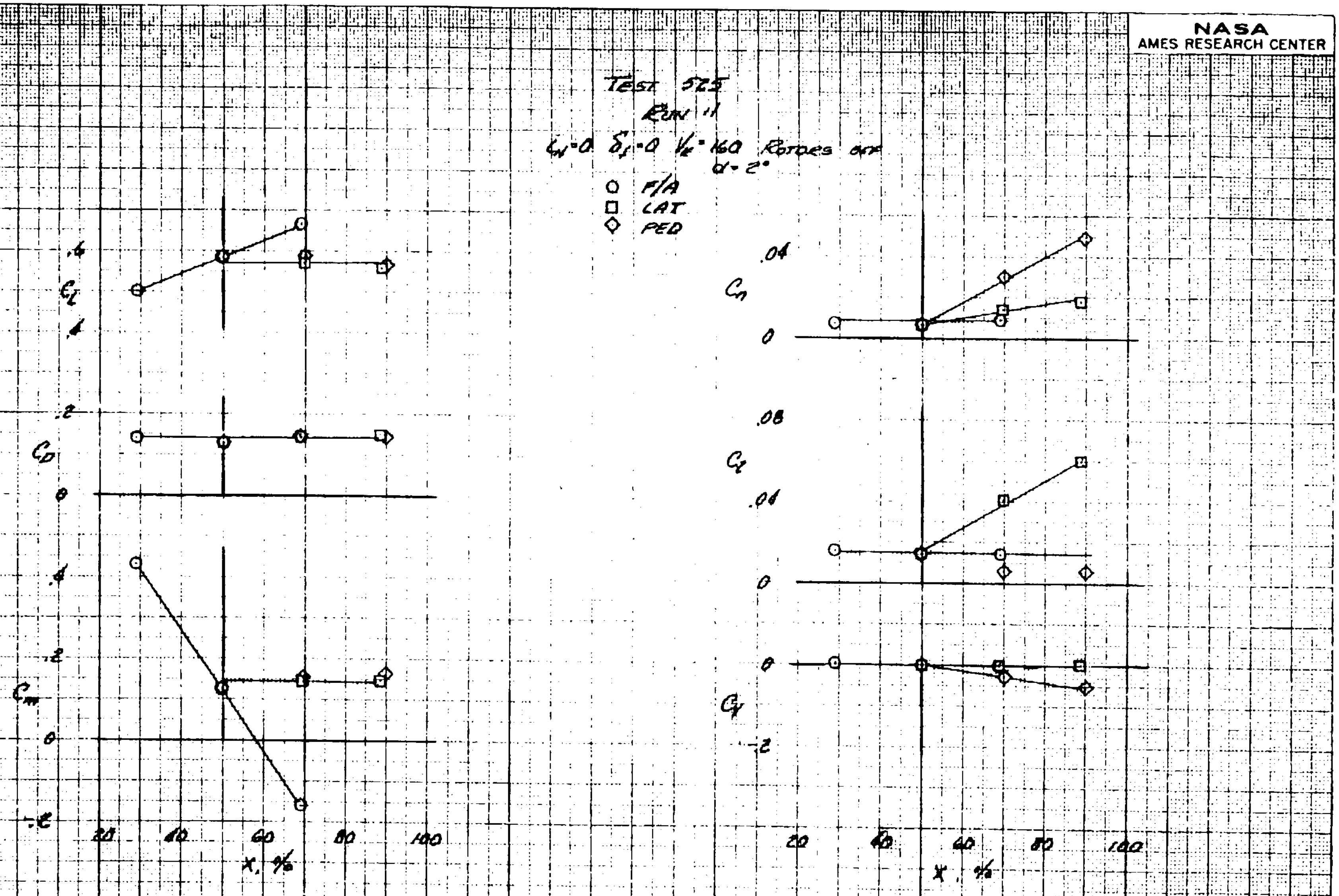


FIGURE 86. – CONTROL EFFECTIVENESS ROTORS OFF.

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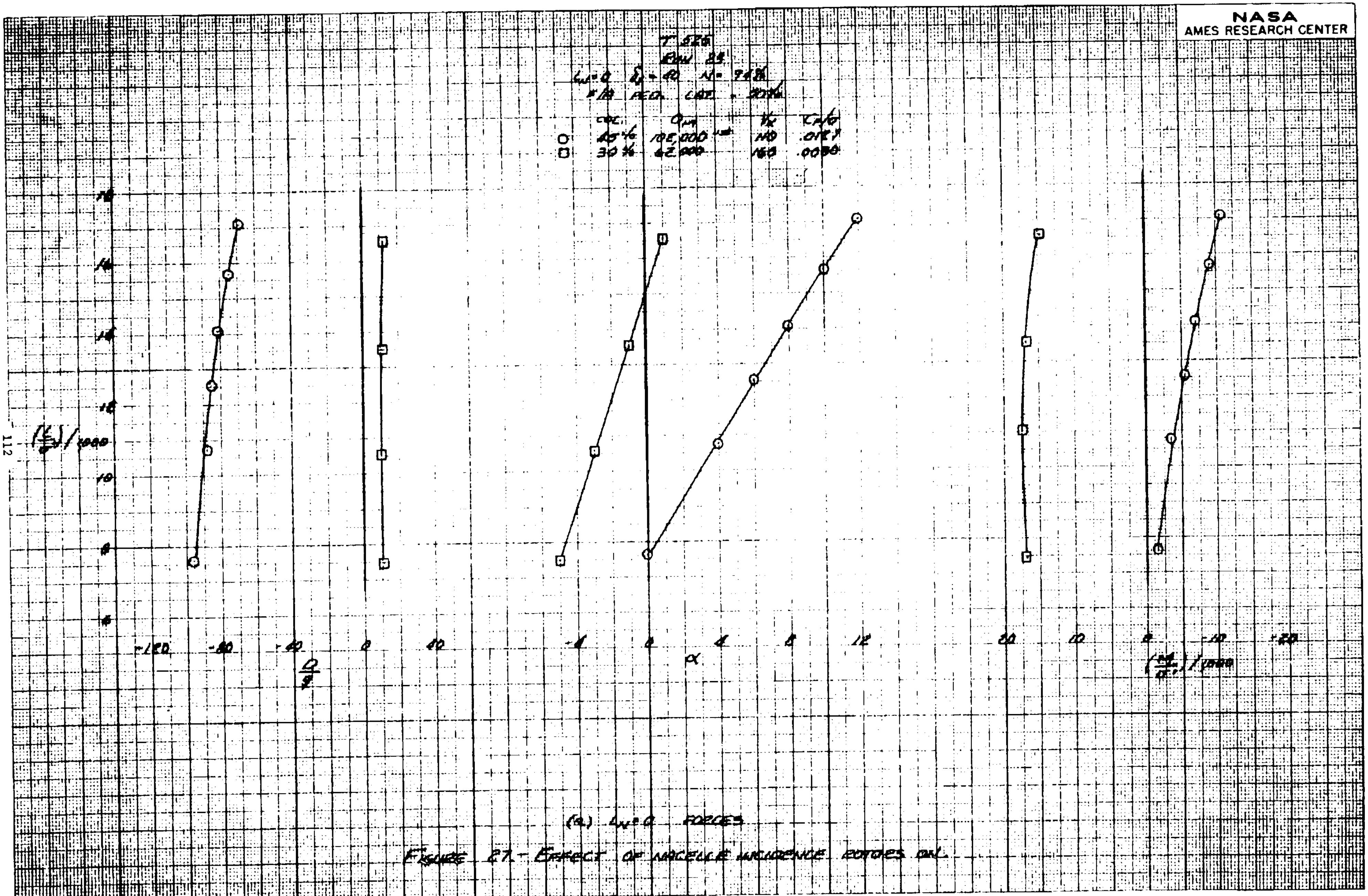
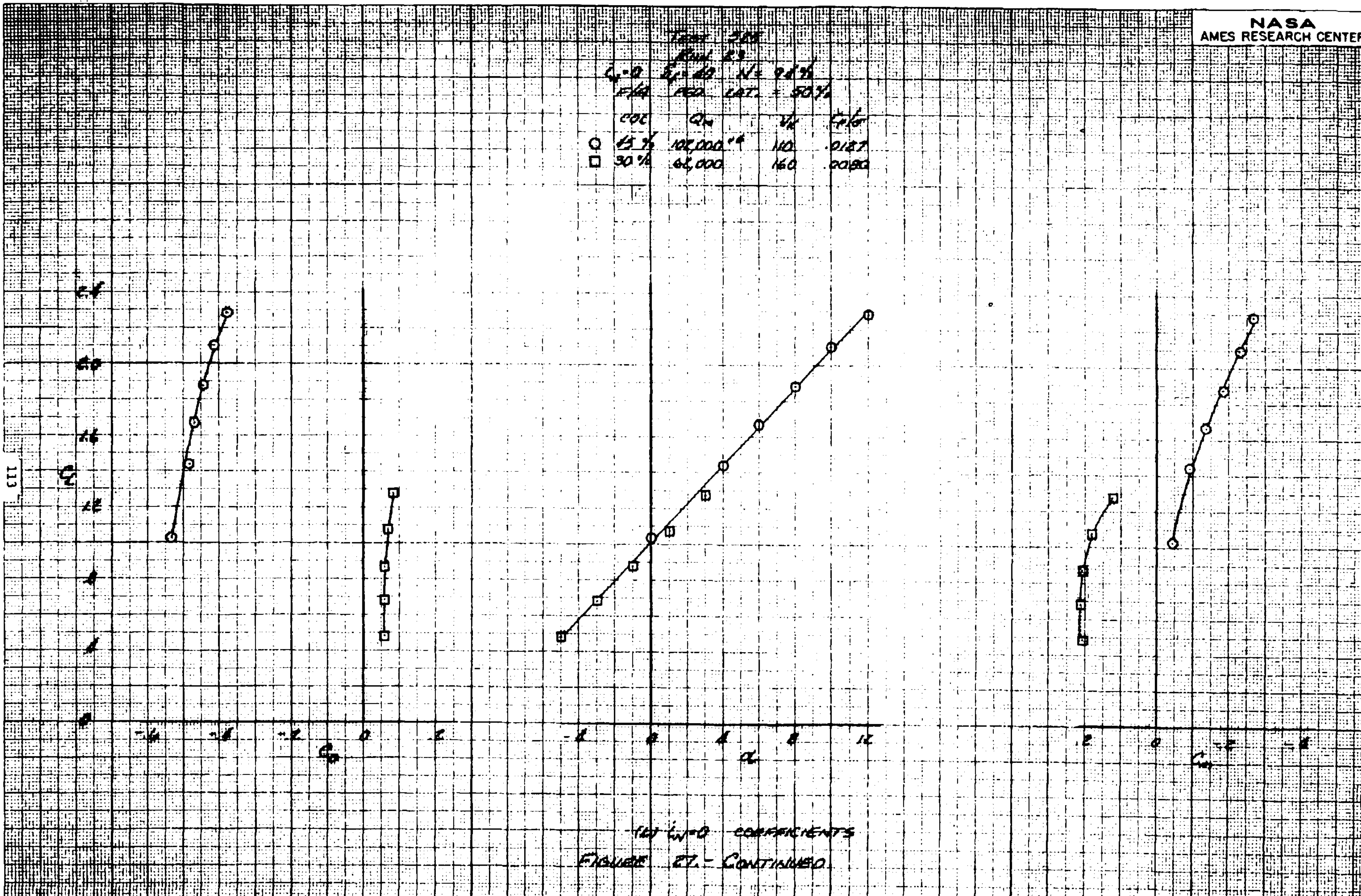


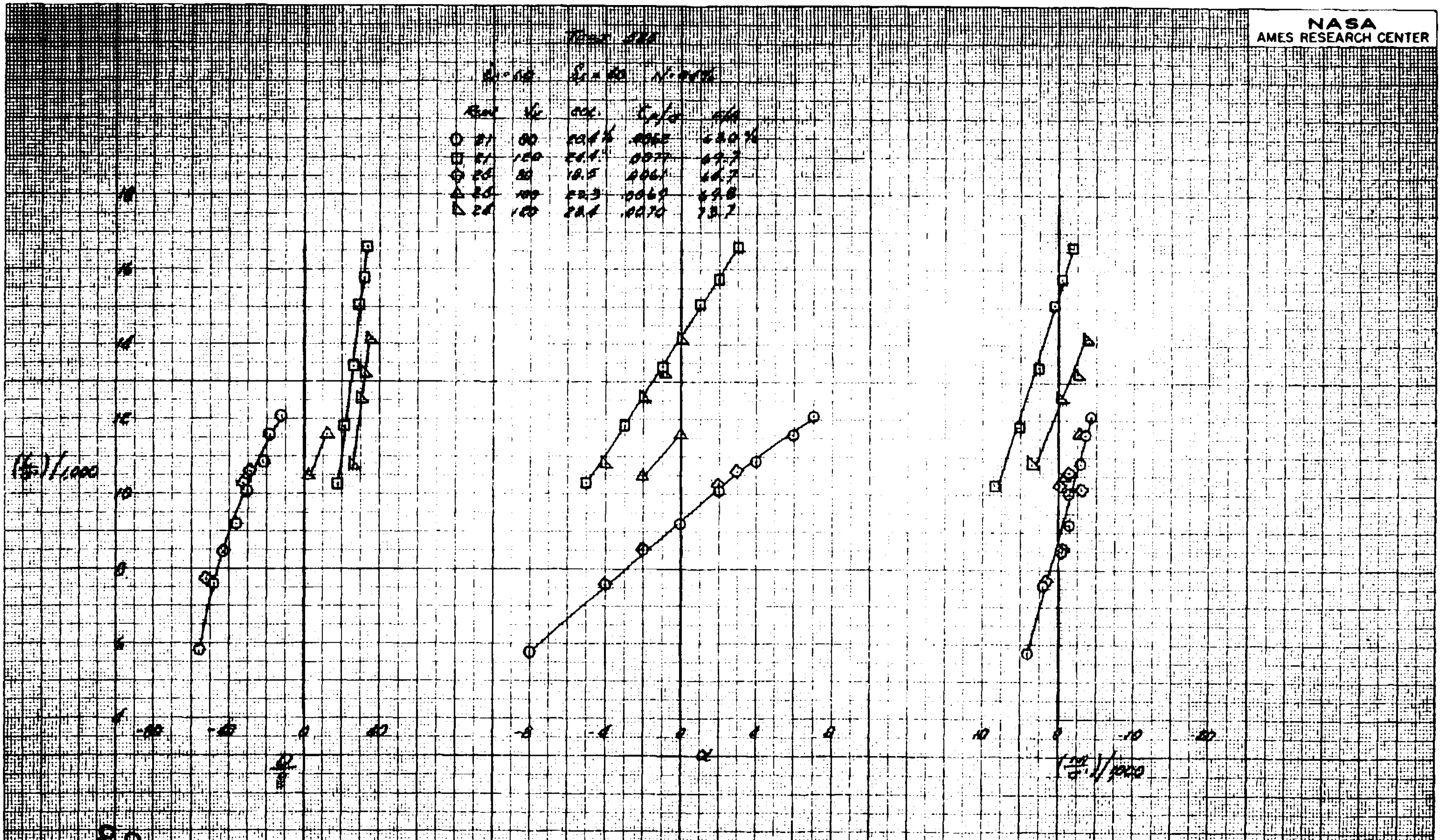
(a) $i_N = 0$ FORCES

FIGURE 27.- EFFECT OF NACELLE INCIDENCE ROTORS ON.



(c) $i_w = 0$ coefficients

Figure 27.- Continued



(c) $i_T = 60°$ Forces

Figure - Continued

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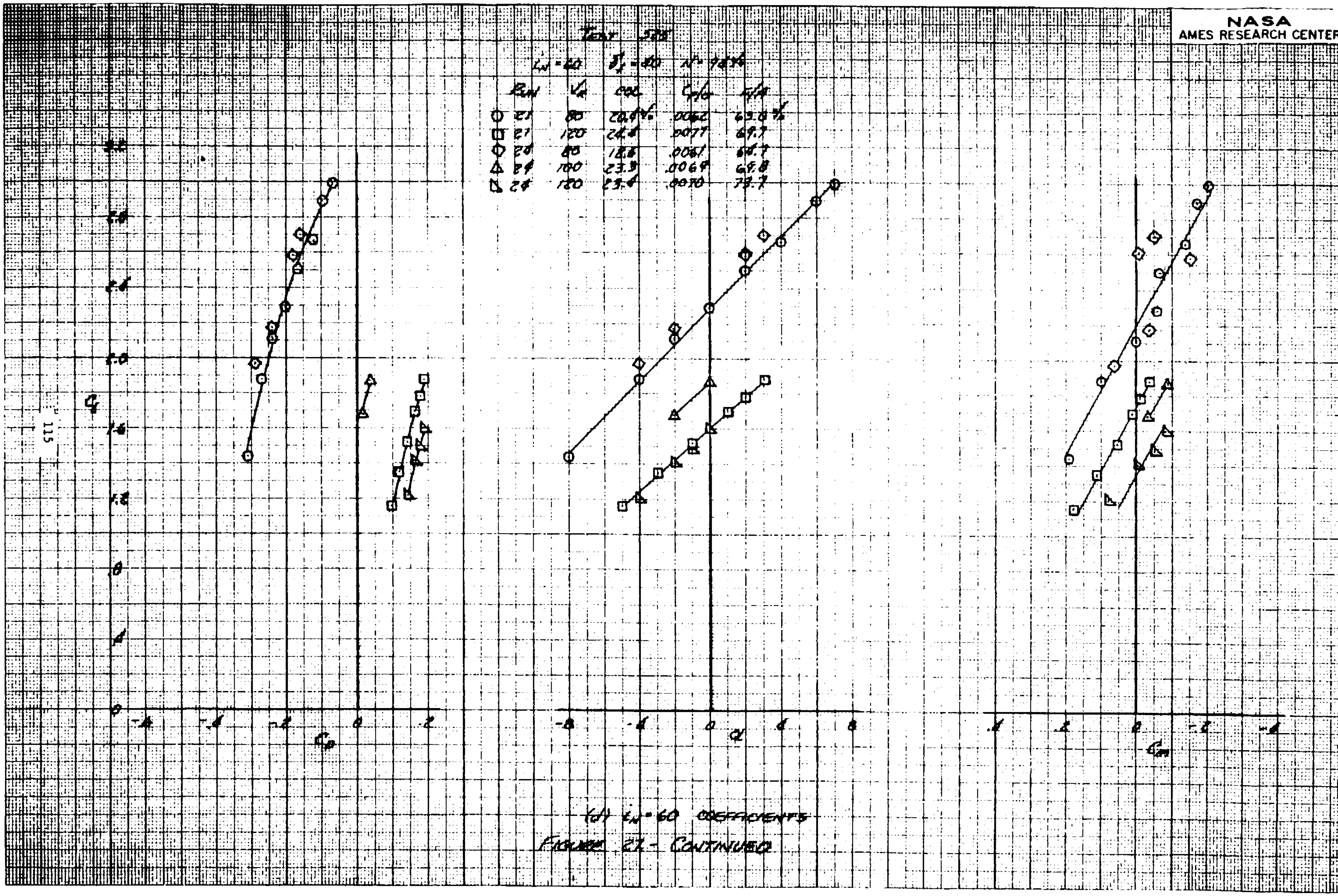
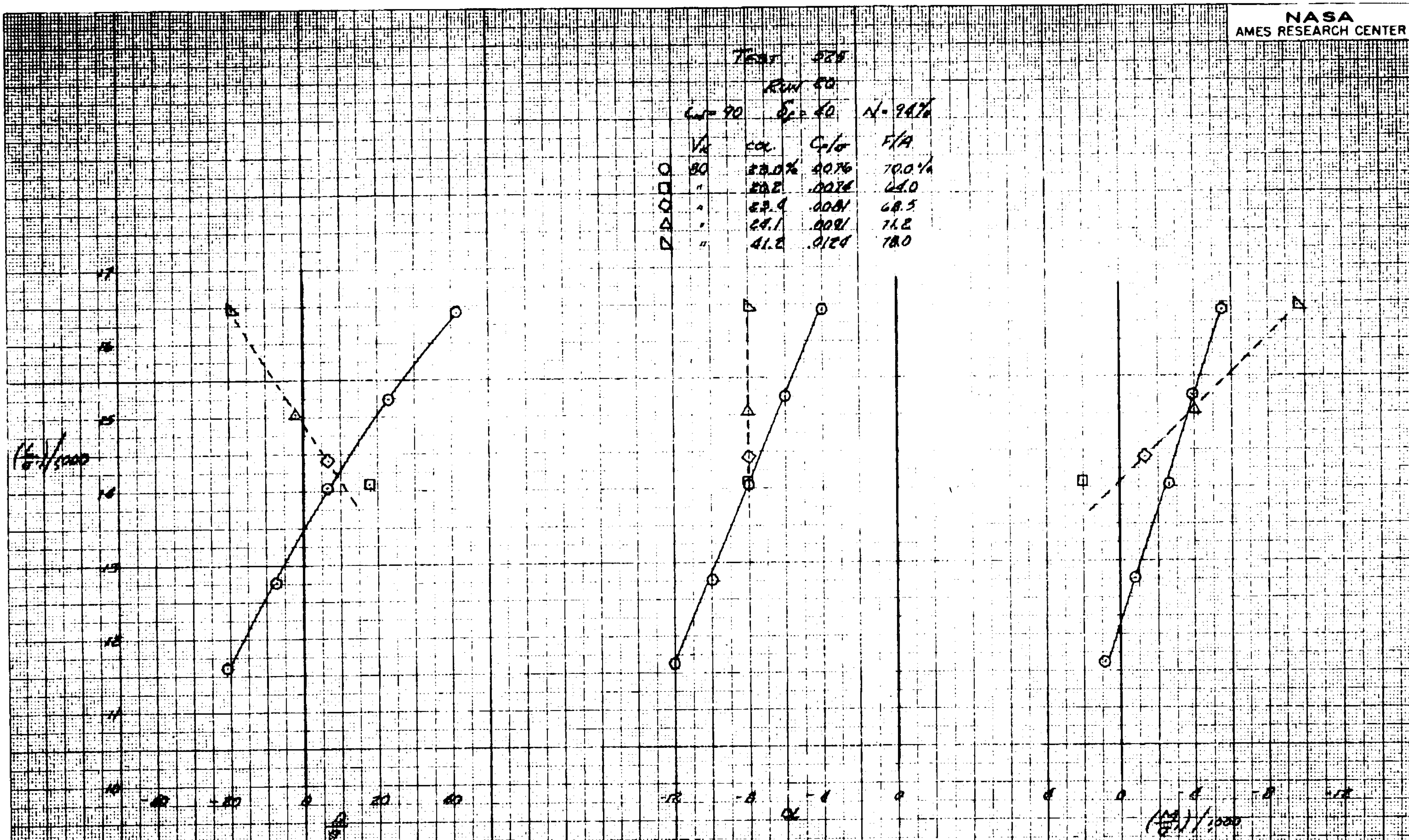


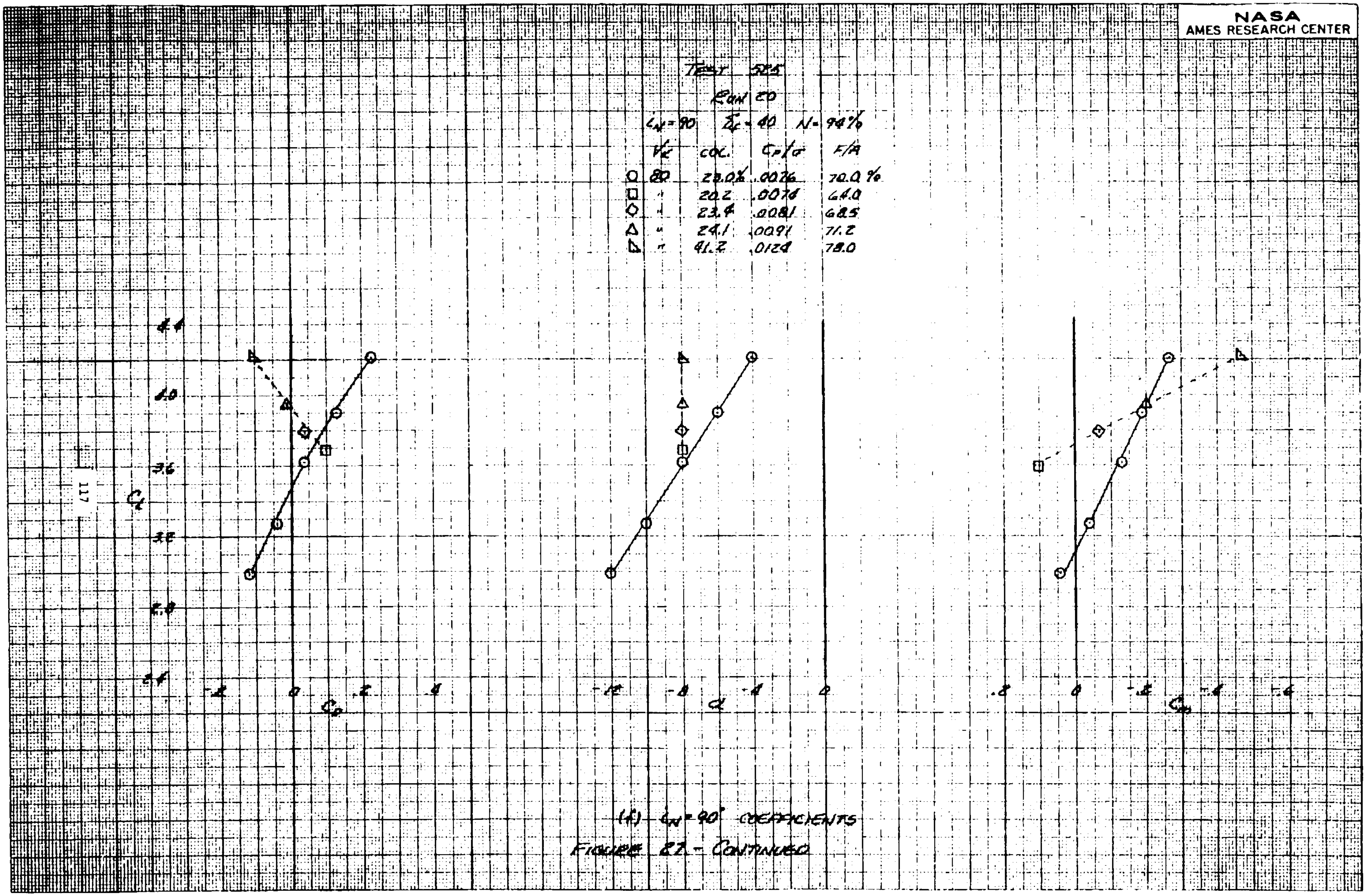
(d) i_N = 60 COEFFICIENTS

FIGURE 27.- CONTINUED



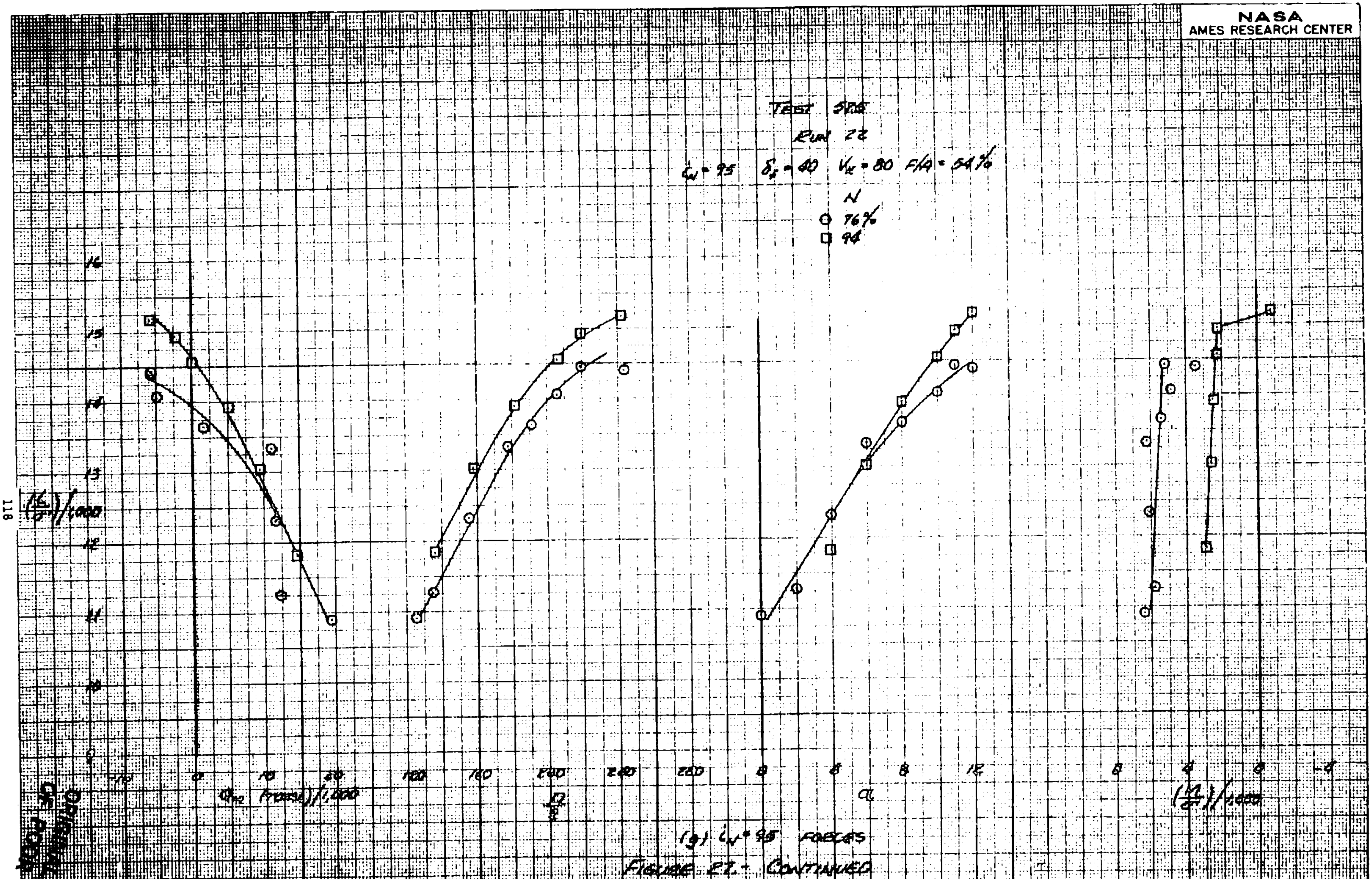
(e) i_N = 90° forces

Figure 27.- Continued



(f) $i_N = 90°$ COEFFICIENTS

FIGURE 22 - CONTINUED



(g) $i_N = 95$ FORCES

FIGURE 27.- CONTINUED

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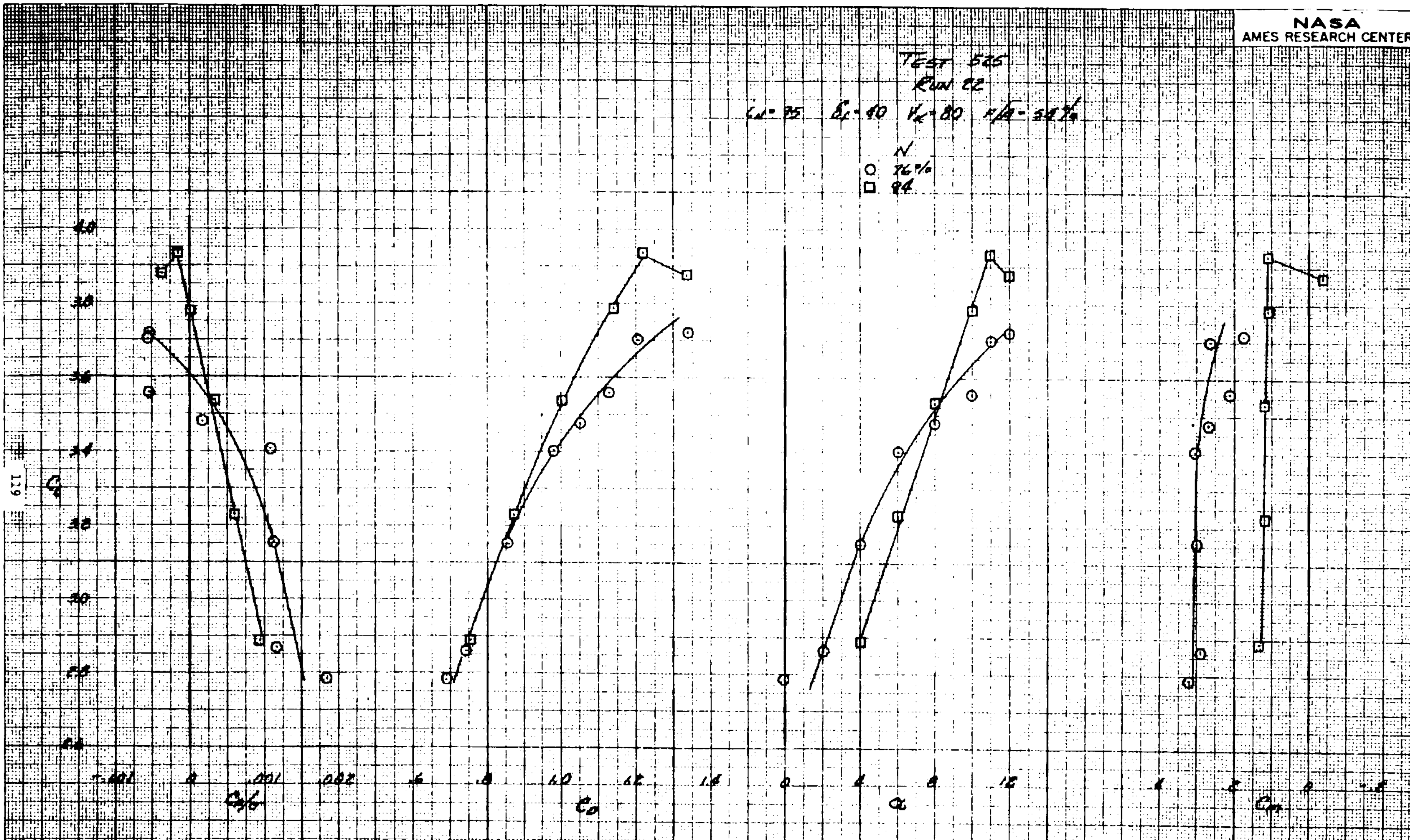


(h) C_N = 95 COEFFICIENTS

FIGURE 27 - CONTINUED.

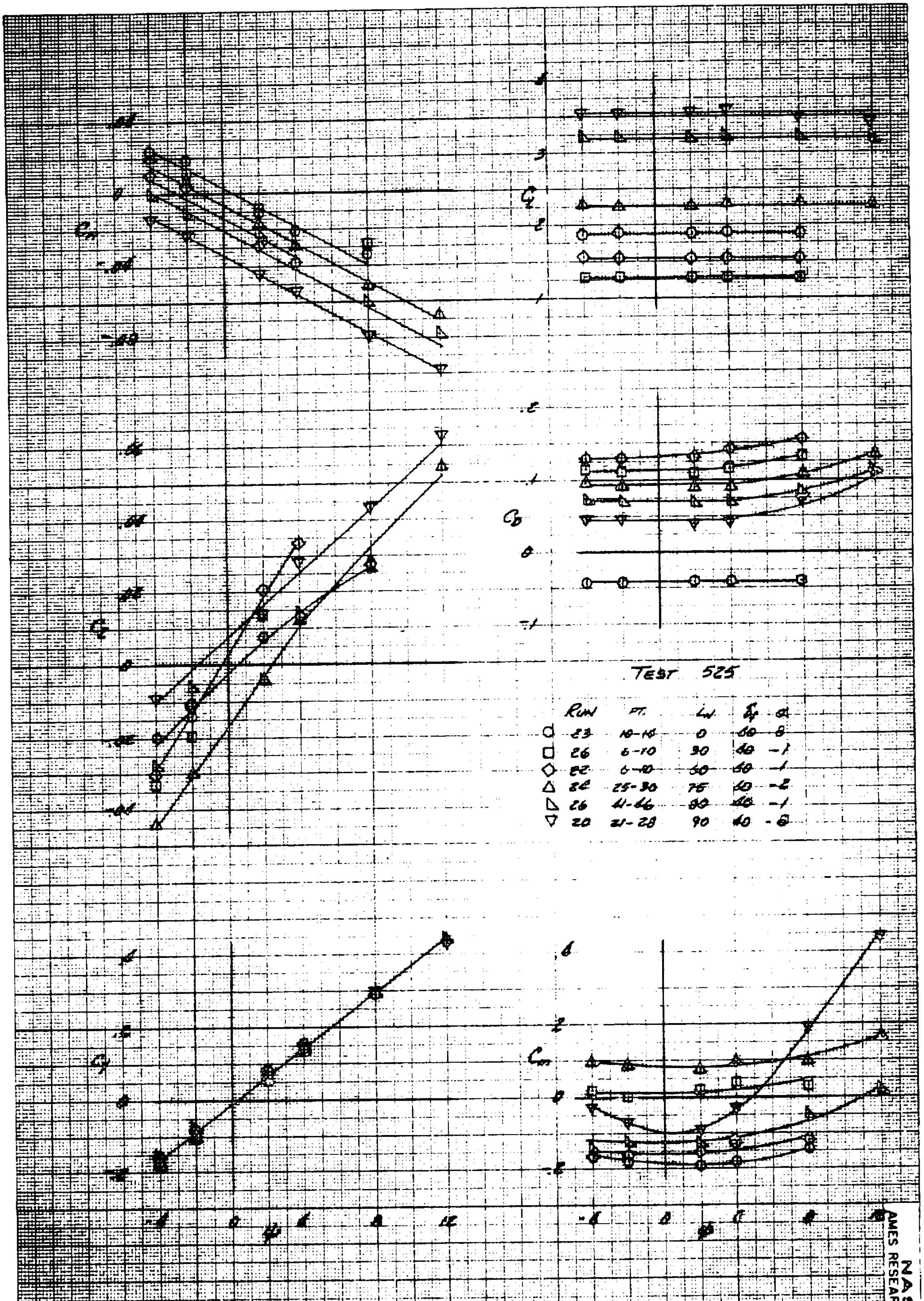
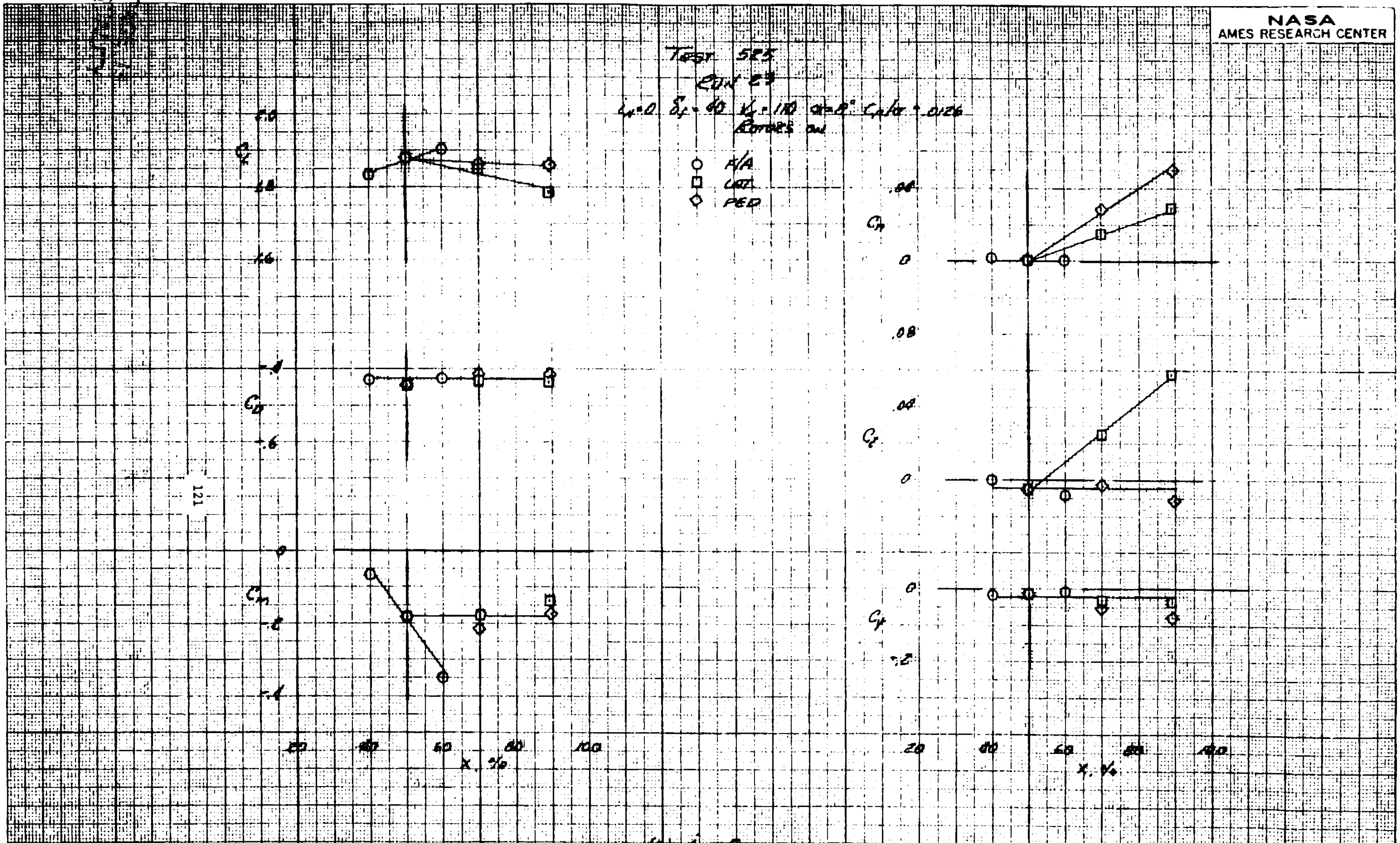


FIGURE 28.- CHARACTERISTICS IN YAW, ROTORS ON.



(a) $C_N = 0$

FIGURE 29.- CONTROL EFFECTIVENESS, ROTORS ON.

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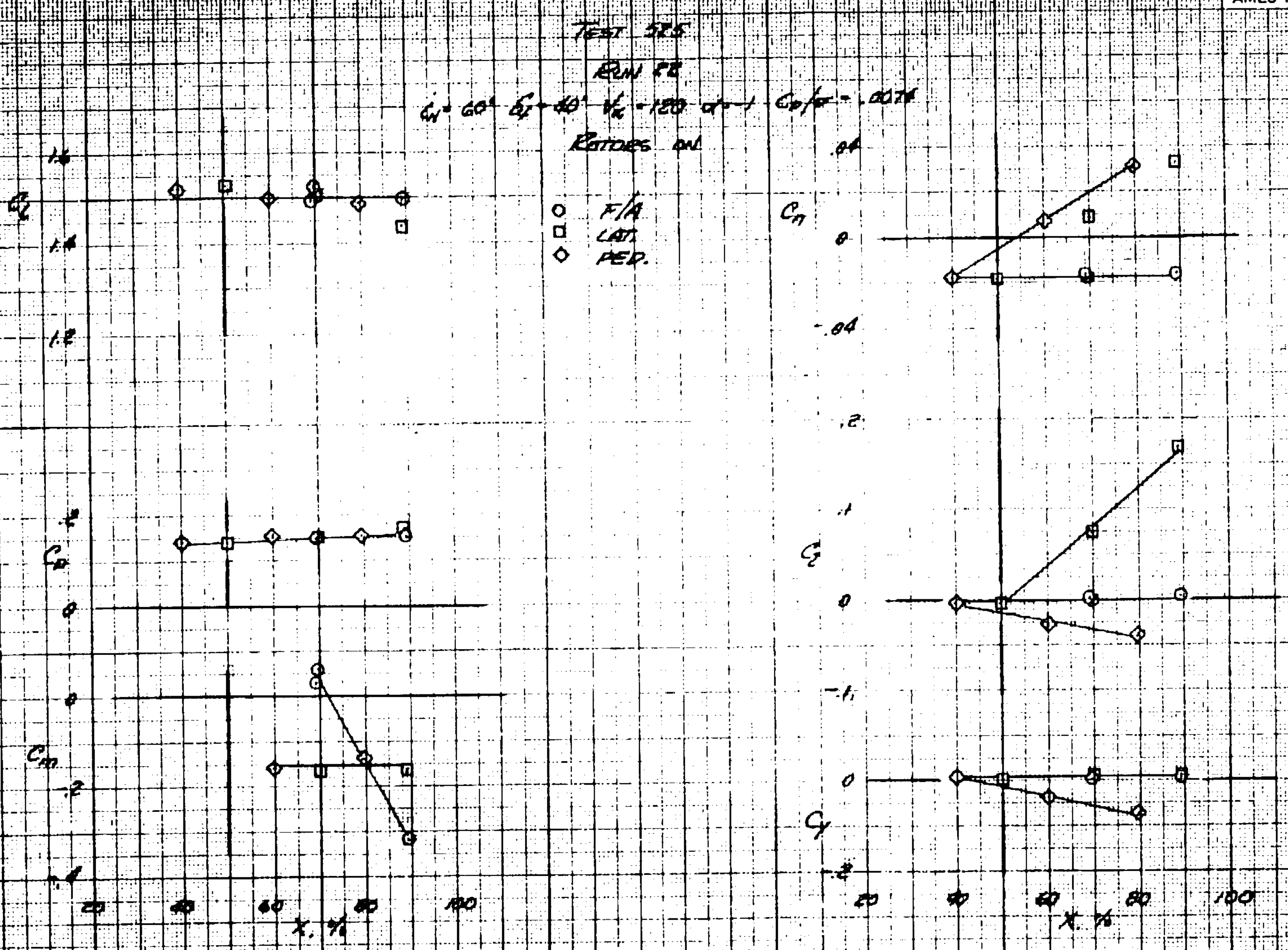
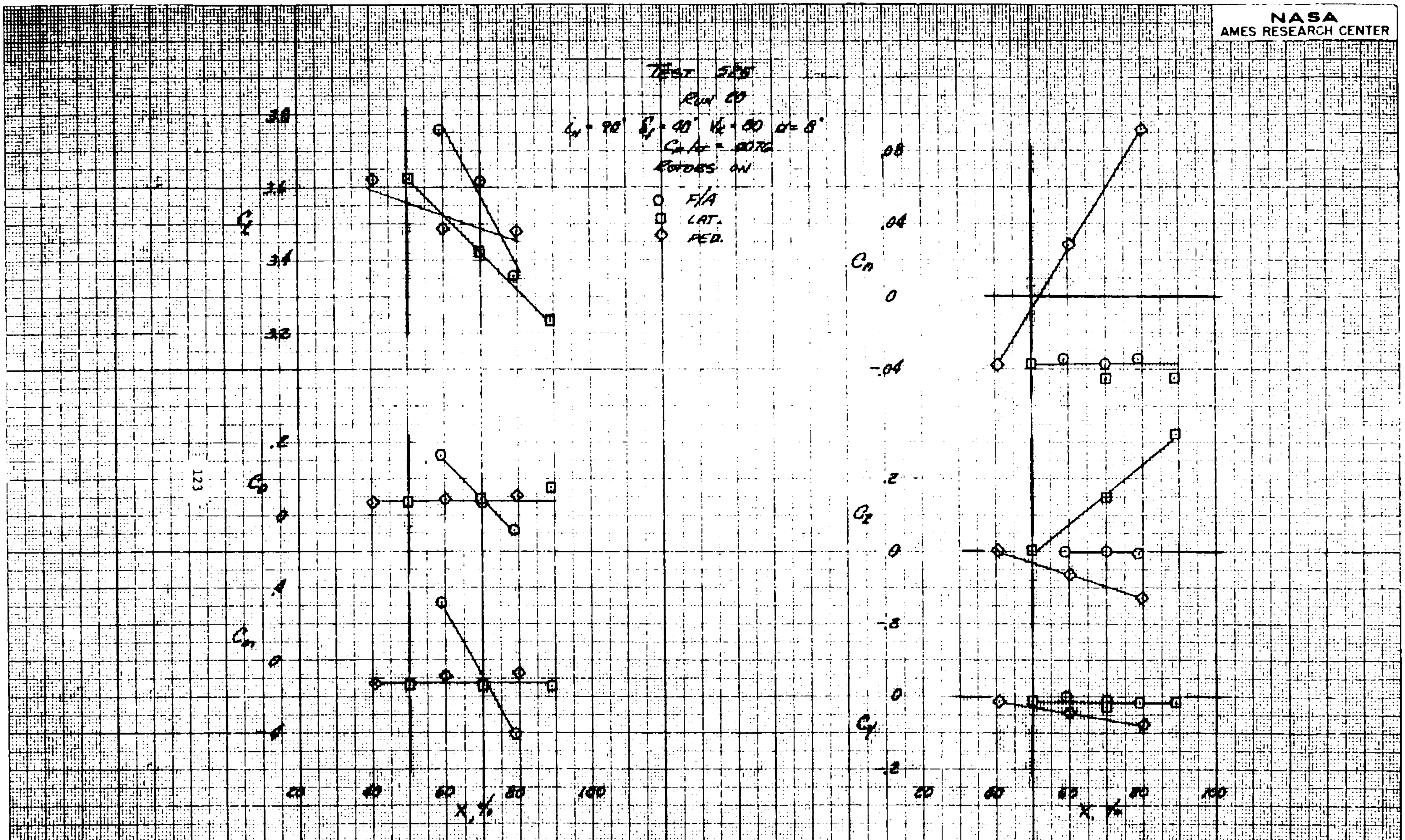


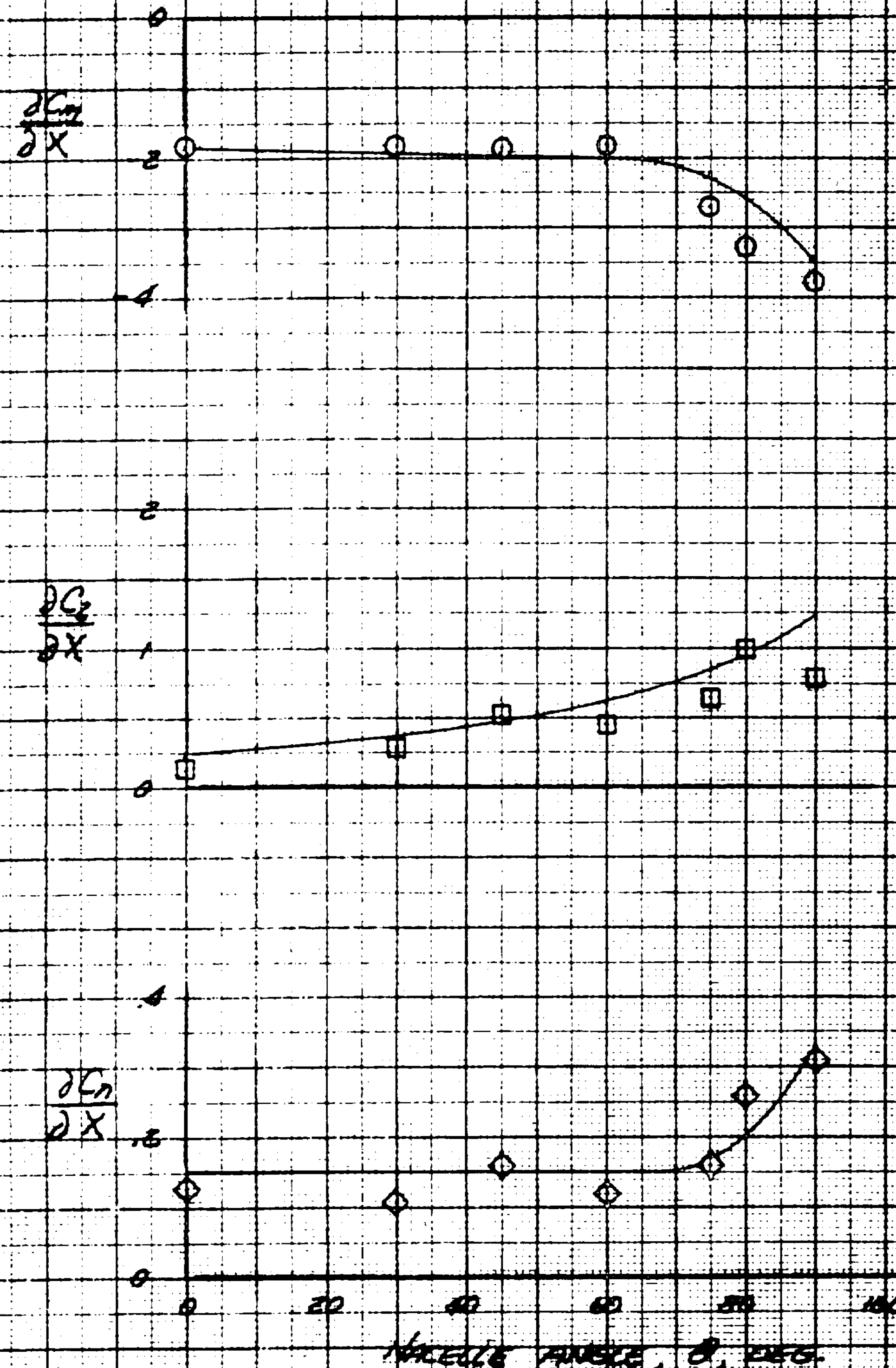
(a) C_N = 60

Figure 29.- Continued



(d) $i_w = 90$

FIGURE 59.- CONTINUED.



(d) VARIATION WITH i_N

FIGURE 29 - CONCLUDED.

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	TEST	RUN	SCALE	CONFIG	SUPPORT STRUTS
▲	ARC 525	10	FULL	V.G. @ .25c	3
◆	"	6	"	V.G. OFF	3
▲	"	51	"	TAPED R/C OFF	3
○	LSWT 564	116	1/5	CLEAN	3
□	"	113	"	CLEAN	1

ROTORS OFF, $i_N = 0$, $\delta_f = 0$

NO TARE OR INTERFERENCE CORRECTIONS

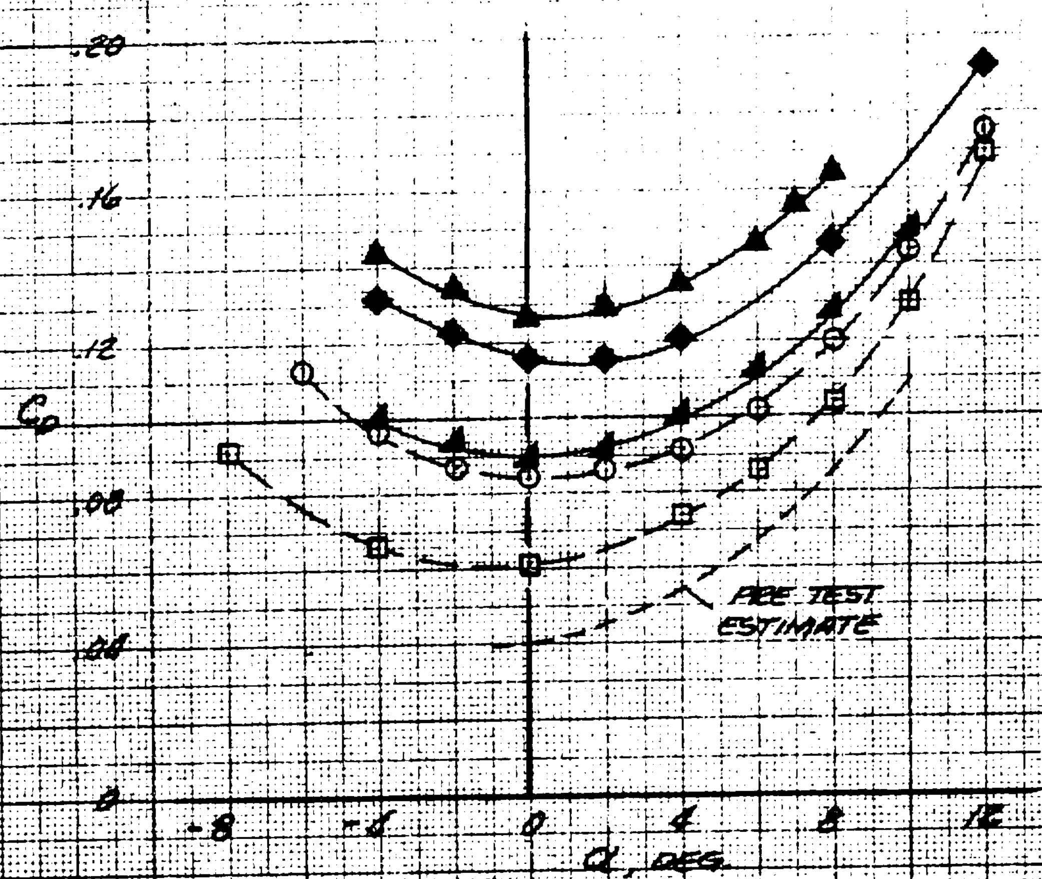
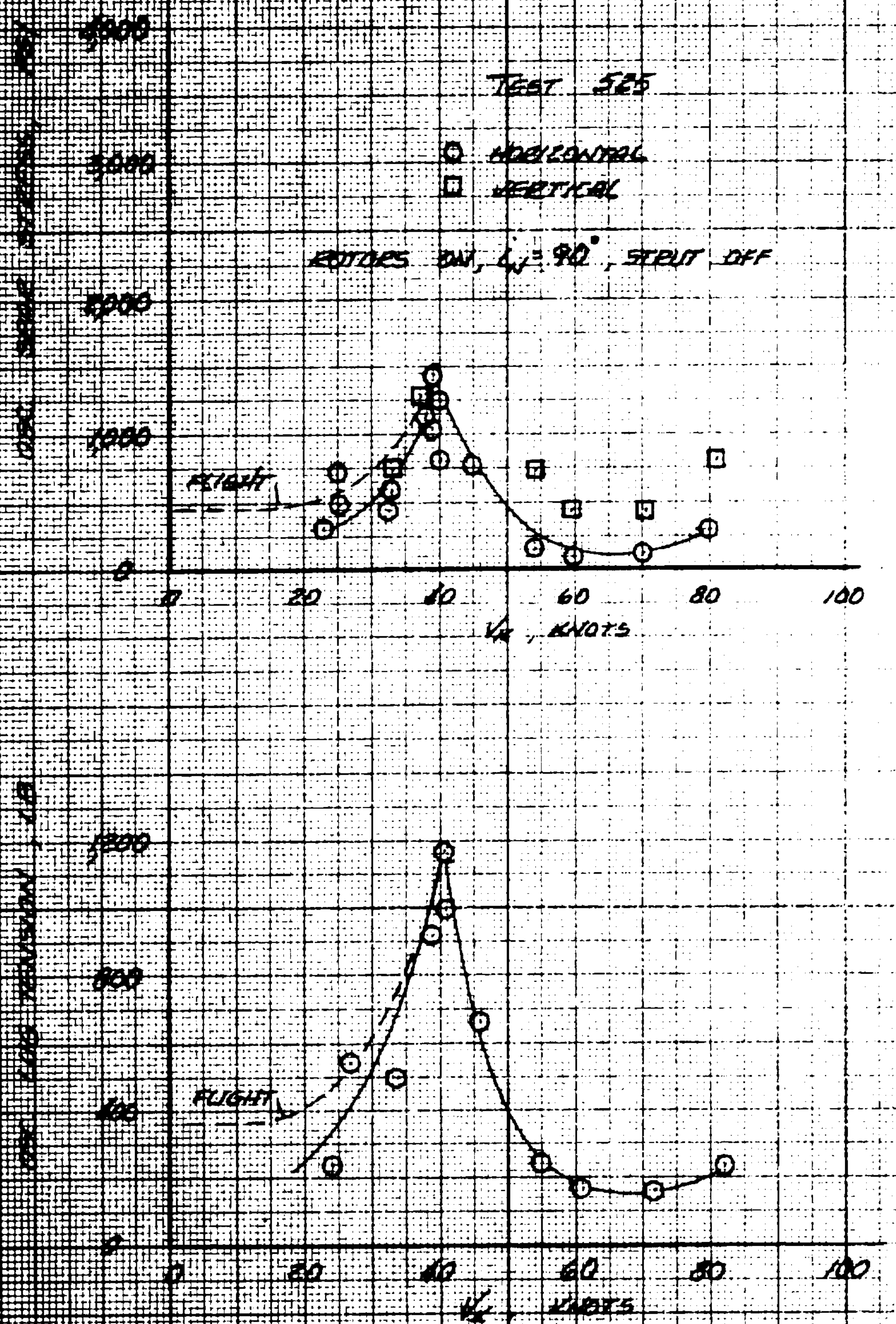


FIGURE 30 - DRAG SUMMARY.



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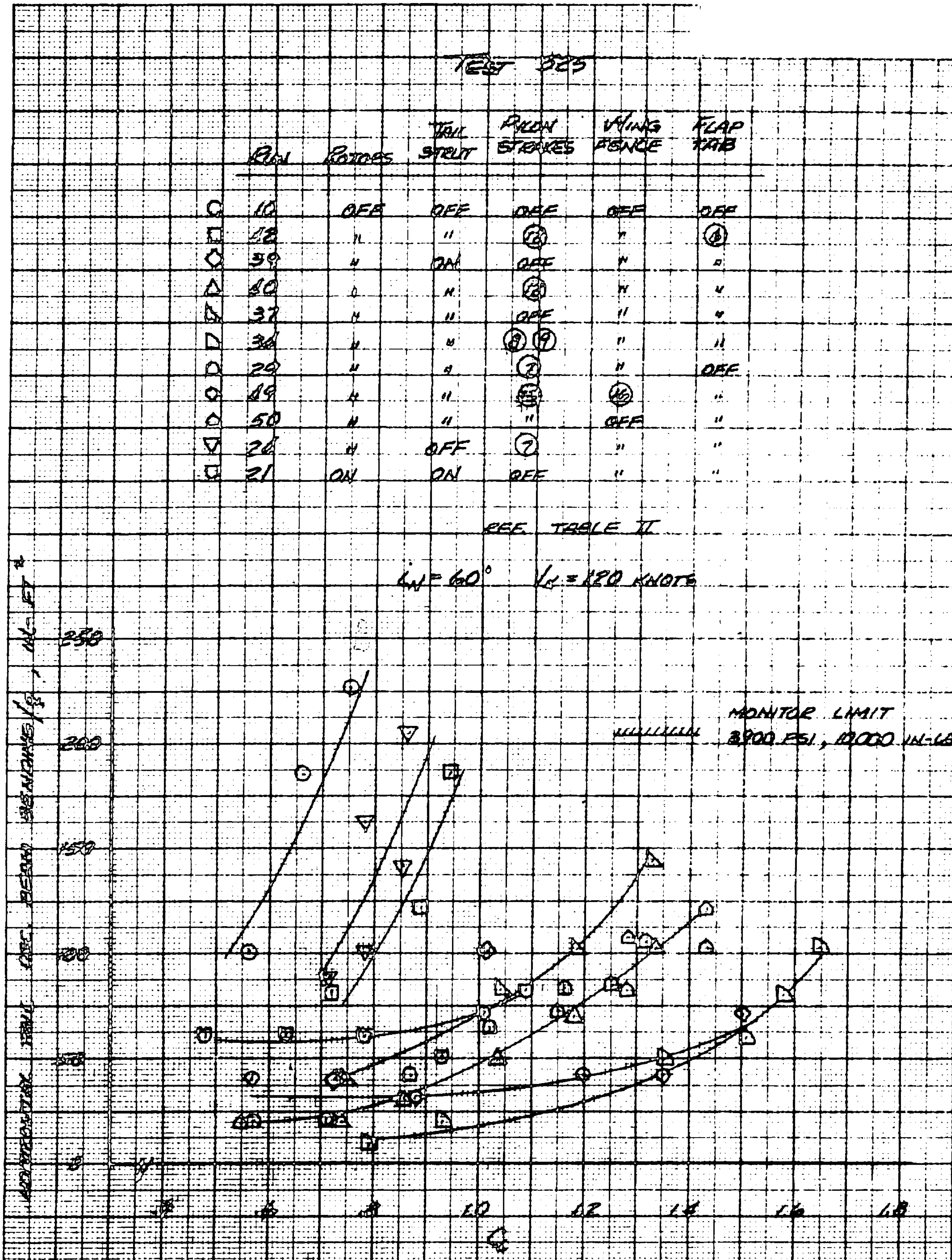
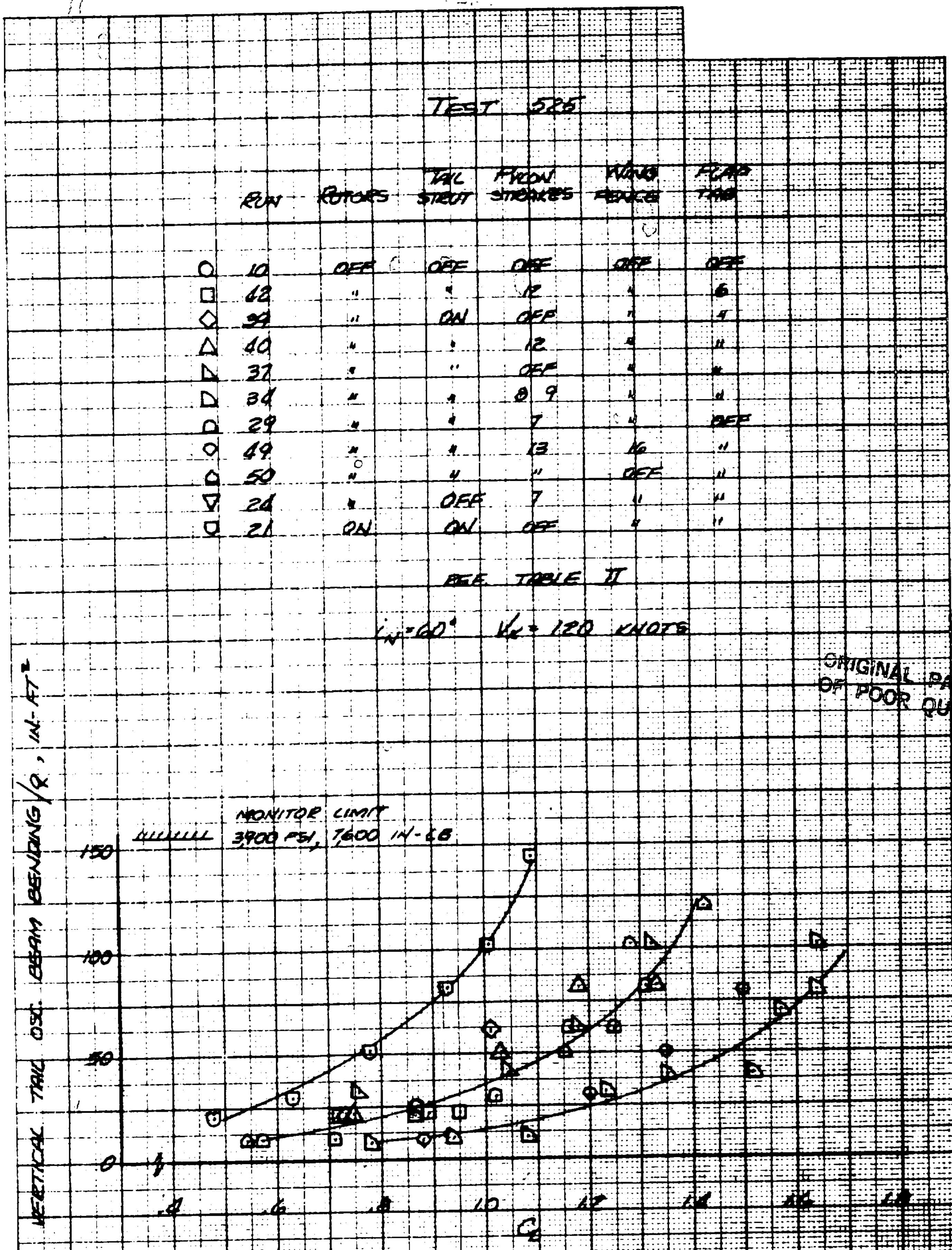
(a) HELICOPTER MODE.

FIGURE 81 - EMPENNAGE LOADS



(b) TRANSITION MODE, HORIZONTAL TAIL.

FIGURE 31.- CONTINUED.



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(c) TRANSITION MODE, VERTICAL TAIL

FIGURE 31 - CONCLUDED.

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16. Abstract

The XV-15 Tilt Rotor Research Aircraft was tested in the Ames 40- by 80-Foot Wind Tunnel for preliminary evaluation of aerodynamic and aeroelastic characteristics prior to flight. The tests were undertaken to investigate the aircraft performance, stability, control and structural loads for flight modes from helicopter through transition and airplane mode up to the tunnel capability of 170 knots. Results from these tests are presented.

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