

SOLUTION OF ENFORCED BOUNDARY
MOTION IN DIRECT TRANSIENT AND
HARMONIC PROBLEMS

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INTRODUCTION

The current versions of NASTRAN, i.e., NASA, MSC, and MAC support non-zero boundary displacements only in the static analysis. Forcing functions in the dynamic analysis formats allow only forces and pressures to exercise the mathematical model. This limitation can be circumvented by the application of a DMAP alter sequence. For the direct harmonic problem, a simple change to module FRRD can be easily incorporated to effect a more efficient use of the code.

Let the equation of motion be written with the dynamic set of coordinates in partition form with subscript b as the boundary set and subscript c as the complimentary boundary set, i.e.,

$$\begin{bmatrix} m_{cc} & m_{cb} \\ m_{bc} & m_{bb} \end{bmatrix} \begin{bmatrix} \ddot{X}_c \\ \ddot{X}_b \end{bmatrix} + \begin{bmatrix} d_{cc} & d_{cb} \\ d_{bc} & d_{bb} \end{bmatrix} \begin{bmatrix} \dot{X}_c \\ \dot{X}_b \end{bmatrix} + \begin{bmatrix} k_{cc} & k_{cb} \\ k_{bc} & k_{bb} \end{bmatrix} \begin{bmatrix} X_c \\ X_b \end{bmatrix} = \begin{bmatrix} \bar{P}_c \\ P_b \end{bmatrix} + \begin{bmatrix} P_{nl} \\ 0 \end{bmatrix} \quad (1)$$

where

m, d, k = mass, damping, and stiffness matrix coefficients

P, P_{nl} = linear and non-linear load vectors

Equation (1) is not solved by the direct transient or frequency formats when p_c , X_b , and therefore \dot{X}_b and \ddot{X}_b , are known and P_b , X_c , and therefore \dot{X}_c and \ddot{X}_c are unknown. However, equation (1) can be rewritten in the form needed for solution by the standard NASTRAN modules. The first of these are:

$$[m_{cc}] \ddot{[X]} + [d_{cc}] \dot{[X]} + [k_{cc}] [X_c] = [P_c] + [P_{nl}] \quad (2)$$

where

$$[P_c] = [\bar{P}_c] + [m_{cb}] \ddot{[X_b]} + [d_{cb}] \dot{[X]} + [k_{cb}] [X_b]$$

By the use of the partitioning modules, the submatrices in Equations (1) or (2) are easily formed. By letting the boundary displacement vector be input through the FORCE or DLOAD cards, the force vector is actually identified as $[P_b] = [X_b]$ (or the first or second derivatives).

The formation of the load vector is different for the transient and harmonic cases. These issues will be discussed below. Somewhat independent of the problem is the requirement that the solution vector to be processed by the remaining modules must be of the dimensions of the "d" set. By using once more partitioning vectors and the MERGE module, the solution vector $[X]$, and in the transient case $[\dot{X}]$ and $[\ddot{X}]$, is merged with the boundary vector $[X_b]$ to form the dynamic vector $[X_d]$. With the "d" set solution vectors formed, the remaining DMAP sequence can be executed without NASTRAN knowing the difference.

In the case of harmonic analysis the non-linear force is zero and equation (2) becomes

$$(w^2 [m_{cc}] + iw [d_{cc}] + [k_{cc}]) [X_c] = [P_c] \quad (3)$$

where

$w = \text{circular frequency, } 2 \pi f.$

HARMONIC ANALYSIS

The DMAP alter that was written to partition the matrix equation (1) into the form of equation (2) and then solve the lower order equation (3) is shown in Figure A-1. The following paragraphs discuss the steps involved.

1. FRRD calculates the load vector PDF and exits the module. The parameter ISKP is changed from -1 to a positive number to be transferred to FRRD the second time the module is executed. If the value of ISKP was set to zero, the default value, the module would have been executed normally. A normal

execution would give a solution to equation (1). The FORTRAN listing of module FRRD is shown in Figure A-2. The added code is underlined: Only the subroutines FRRD1A and FRRD1B are executed in this step.

2. The parameter ISKIP is saved for later use.
3. The partition vector DPAR is used to partition the stiffness matrix KDD. The submatrix identification is related to equation (2) by the following:

Figure A-1. DMAP Alter for Harmonic Response

```

ALTER 159.159
FRRD CASEXX, USETD, DLT, FRL, GMD, GOD, KDD, BDD, MDD, DIT/UDVF, PSF, PDF, PPF/      (1)
      C, N, DISP/C, N, DIRECT/V, N, LUSETD/V, N, MPCF1/V, N, SINGLE/V, N, OMIT/
      V, N, NONCUP/V, N, FRQSET/V, N, ISKIP=-1/ $
SAVE ISKIP $ (2)
PARTN KDD, DPAR, /KD11, KD21, KD12, KD22/ $ (3)
PARTN MDD, DPAR, /MD11, MD21, MD12, MD22/ $ (4)
PARTN PDF, , DPAR/PD11, PD21, PD12, PD22/C, N, 1 $ (5)
MPYAD KD11, PD21, PD11/P1DF/C, N, 0/C, N, -1/ $ (6)
FRRD CASEXX, USETD, DLT, FRL, GMD, GOD, GOD, KD11, , MD11, DIT/UIDVF, PSF, P1DF, (7)
      PPF/C, N, DISP/C, N, DIRECT/V, N, LUSETD/V, N, MPCF1/V, N, MPCF1/V, N, SINGLE/
      V, N, OMIT/V, N, NONCUP/V, N, FROSET/V, N, ISKIP/ $
MERGE KD11, KD21, KD12, KD22, DPAR, /KDD/ $ (8)
MERGE MD11, MD21, MD12, MD22, DPAR, /MOD/ $ (9)
MERGE UIDVF, PD21, PD22, , OPAR/UDVF/C, N, 1 $ (10)

ENDALTER
CEND

```

Figure A-2. Listing of Module FRRD

LEVEL 2.2.1 (DFC 77)

```

ISN 0002      SUBROUTINE FRRD      00000010
C             00000020
C FREQUENCY AND RANDOM RESPONSE MODULE 00000030
C             00000040
C INPUTS CASECC, USETD, ULT, FRL, GMD, GOD, KDD,
      BCD, MDD, PHIDH, DIT      00000050
C             00000060
C OUTPUTS UDV, PS, PD, MP      00000070
C             00000080
C 8 SCRATCHES      00000090
C             00000100
ISN 0003      INTEGER SINGLE, ONIT, CASECC, USETD, DLT, FRL,
      GMD, GOD, BDD, PHIDH, DIT, 1 SCR1, SCR2, SCR3,
      SCR5, SCR6, UDV, PS, PD, FP, PDD, OPTION      00000120
ISN 0004      INTEGER SCP7, SCRB, NAME&2<, MCB&7>      00000130

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ISN	0006	INTEGER FOL	00000140
ISN	0006	COMMON/APP&2<, MODAL&2<, LUSETD, MULTI, SINGLE, OMIT, NONCUP, FRQSET,	00000150
		<u>1 ISKIP</u>	00000155
		C	00000160
ISN	0007	COMMON/FRRDST/OVF&150<, ICNT, IFRST, ITL&3< IDIT, IFRD, K4DD	00000170
ISN	0008	DATA CASECC, USETD, DLT, FRL, GMD, GOD, KDD, HDD, MOD, PHIDH, DIT/ 1 101, 102, 103, 104, 105, 106, 107, 108, 109, 110, 111/	00000180 00000190
ISN	0009	DATA UDV, PS, PD, PP, PDD/201, 202, 203, 204, 203/	00000200
ISN	0010	DATA SCR1, SCH2, SCR3, SCR4, SCR5, SCR6 /301, 302, 303, 304, 305, 306/	00000210
ISN	0011	DATA SCR7, SSCR8/307.308 /	00000220
ISN	0012	DATA MODA /4HMODA/	00000230
ISN	0013	DATA POL/205/	00000240
ISN	0014	DATA NAME /4HFRRD, 4H /	00000250
		C	00000260
		C	BUILD LOADS ON P SET ORDER IS ALL FREQ. FOR LOAD TOGETHER
		C	00000270
		C	00000280
ISN	0015	<u>IF (ISKIP .GE. 0) GO TO 5</u>	00000281
ISN	0017	<u>NLOAD = ISKIP / 2*16</u>	00000282
ISN	0018	<u>NFREQ = ISKIP - NLOAD/*2**16</u>	00000283
ISN	0019	<u>GO TO 15</u>	00000284
ISN	0020	5 <u>CONTINUE</u>	00000285
ISN	0021	CALL FRPDIA&DIT, FRL, CASECC, DIT, PF, LUSETD, NFREQ, NLOAD, FRQSET, FOL, 1 NOTRD<	00000290 00000300
ISN	0022	1F&MULTI.LT.O.AND.SINGLE,LT.O.AND.OMIT.L.T.O AND.MODAL 1 & 1< .NF. MODA< GO TO 60	00000310 00000320 00000330
		C	00000340
		C	REDUCE LOADS TO D OR H SET
		C	00000350
ISN	0024	CALL FPRU14\$PP.USETD, GMD, MULTI, SINGLE, OMIT, MODAL&1<, PHIDH, PD, 1 PS, SCR5, SCR1, SCR2, SCR3, SCR4<	00000360 00000370
ISN	0025	<u>IF (ISKIP .LT. 0) GO TO 40</u>	00000375
ISN	0027	<u>15 CONTINUE</u>	00000377
ISN	0028	<u>IF (MULTI .LT. 0 .AND.SINGLE.LT.O .AND. OMIT .LT. 0 . .AND. MODAL(1) .NE. MODA) POD = PD</u>	00000378 00000379 00000380
		C	00000390
		C	SCR5 HAS PH IF MODAL FORMULATION
		C	00000400
ISN	0030	IF &MODAL&1< .EQ.MODA< PDD #SCR5	00000410
		C	00000420
		C	SOLVE PROBLEM FOR EACH FREQUENCY
		C	00000430 00000440
ISN	0032	IF&NONCUP .LT. 0 .AND. MODAL&1< .EQ. MODA<	00000450

	GO TO 50	
ISN 0034	10 IF&FREQ .EQ. 1 .OR. NLOAD .EQ 1< SCR6 # UDV	00000460
ISN 0036	DO 20 1#1,NFPEQ	00000470
ISN 0037	CALL KLOCK&LOCK&ITIME1<	00000480
	C	00000490
	C FORM AND DECOMPOSE MATRICES	00000500
	C	00000510
ISN 0038	CALL FRRD1C&FRL,FROSET,MDD,RDD,KDD.1,SCR1, SCR2,SCR3,SCR4,SCR8, 1 SCP7.1GOOD<	00000530
	C	00000540
	C ULL IS ON SCR1 -- LLL IS IN SCR2	00000550
	C	00000560
	C SOLVE FOR PD LOADS STACK ON SCR6	00000570
	C	00000580
	C	00000590
ISN 0039	CALL FRRD1D&PDD,SCR1,SCR2,SCR3,SCR4,SCR6, NLOAD,1GOOD,NFREQ<	00000600
ISN 0040	CALL KLOCK&ITIME2<	00000610
ISN 0041	CALL IMTOGO&ITLEFT<	00000620
ISN 0042	IF&2*&ITIME2-ITIME1<.GT. ITLEFT .AND. I .NE. NFREQ< GO TO 70	00000630
ISN 0044	20 CONTINUE	00000640
ISN 0045	1 # NFREQ	00000650
ISN 0046	30 CONTINUE	00000660
ISN 0047	IF&NFREQ .EQ. 1 .OR. NLOAD .EQ 1< GO TO 40	00000670
	C	00000680
	C RESORT SOLUTION VECTORS INTO SAME ORDER AS LOADS	00000690
	C	00000700
ISN 0049	CALL FRRD1E&SCR6,UDV,NLOAD,I<	00000710
ISN 0050	40 ISKIP = NFREQ +NLOAD*2**16	00000720
ISN 0051	RETURN	00000725
	C	00000730
	C UNCOUPLED MODAL	00000740
	C	00000750
ISN 0052	50 CALL FRRD1F&MDD,HDD,KDD,FRL,FRQSET,NLOAD, NFREQ,PDD,UDV<	00000760
ISN 0053	GO TO 40	00000770
ISN 0054	60 PDD # PP	00000780
ISN 0055	GO TO 10	00000790
	C	00000800
	C INSUFFICIENT TIME TO COMPLETE ANOTHER LOOP	00000810
ISN 0056	70 CALL MESSAGE&.5.NFREQ-I,NAME<	00000820
ISN 0057	MCA&1< # SCR6	00000830
ISN 0058	CALL RDTFL&MCA*1<<	00000840
ISN 0059	MDONE # MCD&2<	00000850
ISN 0060	MCR&1< # PP	00000860
ISN 0061	CALL ROTR1&MCH&1<<	00000870
ISN 0062	MCR&2< NOONF	00000880
ISN 0063	CALL WRT1FL&MCB&1	00000890

ISN 0064	IF&SINGLE .LT. 0< GO TO 80	00000900
ISN 0066	MCA&1< # PS	00000910
ISN 0067	CALL PUTRL&MCA&1<<	00000920

$$\begin{aligned}
 K_{dd} &= KD11 \\
 K_{cb} &= KD12 \\
 K_{bc} &= KD21 \\
 K_{bb} &= KD22
 \end{aligned}$$

4. The partition of the mass matrix, MDD, is similar to the stiffness matrix.
5. Because the load vector is calculated for all frequencies and loading conditions at once, PDF is a load matrix, a load vector in each column. The partition vector DPAR is used again to separate the enforced displacements from the forces. The relationship to equation (2) is

$$\begin{aligned}
 P &= PD11 \\
 P_b^c &= PD21
 \end{aligned}$$

6. The module MPYAD performs the matrix multiplication and additions required by equation (2). Here

$$P_c = P1DF$$

7. Module FRRD is executed again, but this time the parameter ISKIP is positive. A jump to statement 15, underlined in Figure A-2, causes only the subroutines FRRD1C, FRRD1E and FRRD1F to be executed. The solution to equation (3) is obtained in this step. The code uses the following names related to equation (3).

$$\begin{aligned}
 M &= MD11 \\
 K^{cc} &= KD11 \\
 P^c &= P1DF \\
 X_c^c &= U1DF
 \end{aligned}$$

8. The stiffness matrices are merged to form the system stiffness matrix. This is the inverse of operation 3.
9. Similar to the stiffness matrix, this operation is the inverse of operation 4.
10. Merges the solution vector X_c of equation (6-7) with X_b to form the system solution vector X_d .

The three merges, operations 8, 9, and 10, are made necessary because NASTRAN uses the displacement approach to the problem solution. In order to calculate stress and forces in the members, the solution vector must contain all grid points.

TRANSIENT ANALYSIS

The DMAP Alter required for the Rigid Format 9, Direct Transient Response, is shown in Figure A-3. The discussions below relates to the circled numbers in the DMAP listing.

1. The Stiffness matrix is partitioned in accordance with Equation (2) where

$$\begin{aligned}
 KD11 &= K_{cc} \\
 KD12 &= K_{cb} \\
 KD21 &= K_{bc} \\
 KD22 &= K_{bb}
 \end{aligned}$$

2. The Mass matrix is partitioned similar to the Stiffness matrix

$$\begin{bmatrix} \text{MDD} \end{bmatrix} = \begin{bmatrix} \text{MD11} & | & \text{MD12} \\ \text{-----} & | & \text{-----} \\ \text{MD21} & | & \text{MD22} \end{bmatrix}$$

Figure A-3. DMAP Alter to Rigid Format 9

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ALTER 163
PARTN KDD,OPAR,/KD11,KD21,KD12,KD22/ $ (1)
PARTN MDD,OPAR,/MDLL,MD21,MD12,MD22/ $ (2)
PARTN PD,OPAR/PD11,PA21,PD12,PD22/C,N,1 $ (3)
MPYAD PA21,MV1,/PBT21/C,N,O/C,N,1/C,N,O/C,N,2 $ (4) (5)
ADD PBT21,PA21/PB21/C,Y,ALPHA=(0.550E-2.0)/C,Y,BETA=(0.550E-2.0)$ (6)
MPYAD PB21,MAIT,/PV21/C,N,O/C,N,1/C,N,O/C,N,2 $ (7)
MPYAD PV21,MV1,/PCT21/C,N,O/C,N,1/C,N,O/C,N,2 $ (8)
ADD PCT21,PV21/PC21/C,Y,ALPHA=(0.550E-2.0)/C,Y,BETA=(0.550E-2.0)$ (9)
MPYAD PC21,MAIT,/PU21/C,N,O/C,N,1/C,N,O/C,N,2 $ (10)
MPYAD KD12,PU21,PD11/P1D/C,N,O/C,N,1 /$ (11)
ALTER 165,165
TRD CASEXX,TRL,NLFT,DIT,KD11,MD11,PID/UIDVT,PILD/C,N,DIRECT/
V,N,NOUE/V,N,NONCUP/V,N,NCOL $ (12)
ALTER 166
MERGE KD11,KD21,KD12,KD22,OPAR/KDD/ $ (13)
MERGE MD11,MD21,MD12,MD22,OPAR,/MDD/ $ (14)
MERGE PD11,PILD,PD12,PD22,,OPAR/PNLD/C,N,1 $ (15)
PARTN PA21,PVA,/A21,,PDA12,/C,N,1 $ (16)
PARTN PV21,PVA,/V21,,PDA12,/C,N,1 $ (17)
PARTN PU21,PVA,/U21,,PDA12,/C,N,1 $ (18)
MERGE A21,,V21,,PVVA,/PVA21/C,N,1 $ (19)
MERGE PVA21,,U21,,PVUVA,/PUVA21/C,N,1 $ (20)
MERGE UIDVT,PUVA21,,,,DPAR/UDVT/C,N,1 $ (21)
ENDALTER

```

3. The load vector, PD, which is output from module TRLG, is partitioned according to Equation (2), where

$$PD = \{P(t_1)\}, \{P(t_2)\}, \dots$$

$$PD11 = \{\bar{P}_c(t_1)\}, \{\bar{P}_c(t_2)\}, \dots$$

$$PA21 = \{P_b(t_1)\}, \{P_b(t_2)\}, \dots$$

Note that PD is a matrix formed by columns of load vectors, one column for each time step. The matrices PD22 and PD12 are not generated, i.e.

$$PD = \begin{Bmatrix} PD11 \\ PA21 \end{Bmatrix}$$

4. Direct input matrices, MV1 and MALT, are used subsequently to calculate the velocity and displacement matrices from the acceleration matrix. The forms of MV1 AND MALT are

$$MV1 = \begin{matrix} \begin{bmatrix} 0 & 1 & 0 & 0 & \dots & \dots \\ 0 & 0 & 1 & 0 & \dots & \dots \\ 0 & 0 & 0 & 1 & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots \end{bmatrix} & \begin{matrix} \updownarrow \\ M \\ \updownarrow \end{matrix} \end{matrix}$$

← N+2 →

$$MALT = \begin{matrix} \begin{bmatrix} 1 & 1 & 1 & 1 & \dots & \dots \\ 0 & 1 & 1 & 1 & \dots & \dots \\ 0 & 0 & 1 & 1 & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots \end{bmatrix} & \begin{matrix} \updownarrow \\ M \\ \updownarrow \end{matrix} \end{matrix}$$

← N+2 →

The dimensions of both matrices are M X N + 2 where M is the number of coordinates in the b-set and N is the number of time steps.

5. Produces the matrix product

$$\begin{aligned} [PBT21] &= [PA21] * [MV1] \\ &= [\{P_b(t_1)\}, \{P_b(t_2)\}, \dots] \begin{bmatrix} 0 & 1 & 0 & 0 & \dots & \dots \\ 0 & 0 & 1 & 0 & \dots & \dots \\ 0 & 0 & 0 & 1 & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots \end{bmatrix} \\ &= [0, \{P_b(t_1)\}, \{P_b(t_2)\}, \dots] \end{aligned}$$

It is seen that this operation moves the columns of the acceleration

vectors from time t_i to t_{i+1} .

6. Produces the matrix sum

$$[PB21] = \alpha [PBT21] + \beta [PAZ1]$$

The coefficients α and β are set equal to one-half of the integration time step, Δt .

$$[PB21] = \frac{\Delta t}{2} [\{P_1 + P_2\}, \{P_2 + P_3\}, \dots]$$

$$= [\{\Delta V_1\}, \{\Delta V_2\}, \dots]$$

where $[P_i] = \{P_c(t_i)\}$; $i = 1$ to $N + 2$

The matrix PB21 represents the change in velocity, ΔV_i , between time steps, t_i and t_{i+1} . The calculation is based on the trapezoidal rule for numerical integration.

7. The final step in producing the matrix of velocity vectors, PV21 from the matrix of acceleration vectors, PA21, this module produces the matrix product

$$[PV21] = [PB21] [MAIT]$$

$$= [\Delta V_1 \} , \{ \Delta V_2 \} , \dots]$$

$$= [\{ \Delta V_1 \} , \{ \Delta V_1 + \Delta V_2 \} , \{ \Delta_1 + \Delta_2 + \Delta V_3 \} , \dots]$$

$$\begin{bmatrix} 1 & 1 & 1 & 1 & \dots \\ 0 & 1 & 1 & 1 & \dots \\ 0 & 0 & 1 & 1 & \dots \\ 0 & 0 & 0 & 1 & \dots \\ \vdots & \vdots & \vdots & \vdots & \ddots \\ \vdots & \vdots & \vdots & \vdots & \ddots \end{bmatrix}$$

8., 9. A repeat of operations e, f, g. The matrix of displacement and 10. PU21, is calculated from the matrix of velocity vectors, PV21.

11. The load vector is calculated in accordance with Equation (2).

$$KD12 = K_{cb}$$

$$PU21 = \{X_b\}_1, \{X_b\}_2, \dots$$

$$PD11 = \{\bar{P}_b\}_1, \{\bar{P}_b\}_2, \dots$$

$$P1D = \{P_c\}_1, \{P_c\}_2, \dots$$

12. The module TRD calculates the solution to Equation (2).

$$KD11 = [Kcc]$$

$$MD11 = [Mcc]$$

$$P1D = P_1, P_2, \dots$$

$$UIDVT = \begin{bmatrix} X & X & & \\ \cdot & \cdot & & \\ X & X & & \\ \cdot & \cdot & & \\ X & X & \cdot & \cdot & \cdot & \cdot \end{bmatrix}$$

$$[PILD] = \{P_{\eta\ell}\} .$$

The solution vector, UIDVT, is a matrix of displacements, velocity and acceleration vectors for each grid point; a column for each time step.

13. The system stiffness matrix is formed

$$\begin{bmatrix} KD11 & KD12 \\ \hline KD21 & KD22 \end{bmatrix} = [KDD]$$

14. The system mass matrix is formed similar to the operation (13.)

15. The system load vector is formed

$$\begin{bmatrix} PD11 \\ \hline PILD \end{bmatrix} = [PNLD]$$

- 16., 17. Partition the acceleration, PA21, velocity, PV21, and displacement, PU21, matrices to the correct size to be merged with UIDVT.

- 19., 20. These operations merge the solution matrix, UIDVT, with the excitation matrix, PUVA21, to form the final system solution matrix, UDVT.

$$\begin{bmatrix} UIDVT \\ \hline PUVA21 \end{bmatrix} = [UDVT]$$

From this point on, the solution is calculated according to the Standard Rigid Format 9 procedure.