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USE OF COMPUTER-AIDED TESTING IN THE INVESTIGATION OF PILOT RESPONSE TO CRITICAL IN-FLIGHT EVENTS

VOLUME I

Thomas H. Rockwell and Walter C. Giffin Industrial and Systems Engineering

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FINAL REPORT

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USE OF COMPUTER-AIDED TESTING IN

THE INVESTIGATION OF PILOT RESPONSE TO

CRITICAL IN-FLIGHT EVENTS

VOLUME I

Supported By

National Aeronautics and Space Administration

Ames Research Center

Moffet Field, California 94035

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The Ohio State University Research Foundation #713729

Principal Investigators:

Thomas H. Rockwell

and

Walter C. Giffin

FOREWORD

This report is prepared in two volumes. Volume I reports the findings of the research. Volume II contains the Appendices to the final report. The appendices contain detailed documentation of the tools used to conduct the research. This includes a sample set of displays presented to subjects during computer aided testing, a set of experimenter instructions necessary to operate and modify the programs and a table of contents on the 1981 Symposium on Aviation Psychology supported by this grant.

ACKNOWLEDGEMENTS

Alten- restricted e

The investigators on this project owe a great debt of gratitude to the students who worked so hard to make the project a success, the pilots who generously offered their time as subjects, the secretaria' staff who reduced scrawl to text, and NASA personnel who provided both financial and moral support.

The prime movers on computer program preparation were Jeff Lee and Dave Romer. They were responsible for translating verbal descriptions into computer code. Along the way they invented several novel techniques for dynamic presentations and information exchange. Other graduate students who worked on the project from time to time include Kent Brooks, Mark Thomas and Linda Rings. They aided in subject testing and data interpretation.

The graduate research associates were ably assisted by an outstanding cadre of undergraduate engineering students. At different times during the project Tracy Barber, Kathy Brosch, Mike Romeo and Steve Schoenlein were employed. They organized the data, performed statistical analyses and prepared a myriad of graphs, tables and charts. Project typing was performed by Jean Martens assisted by Linda Parsons and Annie Sears.

We are also indebted to our many friends in the OSU Department of Aviation who offered advice and counsel whenever requested. Professors Jack Eggspheuler, Dick Taylor and Stacy Weislogel were especially helpful. Professor Dick Jensen was the one responsible for the First Symposium on Aviation Psychology supported as part of this project.

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We also want to thank Dr. Charles Billings and John Lauber who together with many of their co-workers at NASA Ames offered continuing guidance and encouragement throughout the project.

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Finally we want to express our sincerest appreciation to all of our pilot friends who gave their valuable time to participate as subjects in these experiments.

> Thomas H. Rockwell Walter C. Giffin

EXECUTIVE SUMMARY

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A critical in-flight event (CIFE) is a situation which is unexpected, unplanned, unanticipated and is perceived by the pilot in command to threaten the safety of the aircraft. The CIFE requires pilot judgement beyond routine decision making or preprogrammed decision structure. The safety of the aircraft depends more on pilot cognitive processes than skilled motor performance.

This research extends the results of earlier research on pilot response to CIFE's by using a computer-aided scenario testing system (CAT). The system makes use of an interactive terminal whereby navigation displays, instrument panel displays and assorted textual material are presented by computer graphics. Communication between subject and computer is accomplished by means of the touch sensitive CRT screen. These programs include biographical data, knowledge survey, a variety of diagnostic scenarios, a destination-diversion scenario, an airport ranking exercise and a combined destination diversion/diagnostic test with dynamic state presentation and control (PLATO-GAT). A complete time history of all data inquiries and responses is maintained for each individual subject tested.

Research Objectives

The objectives of this research were to:

- Design and implement a computer aided testing device for studying pilot diagnosis and destination-diversion decision making.
- 2) Develop new scenarios to take advantage of the capabilities of PLATO⁽⁰⁾.
- 3) Test a variety of candidate hypotheses concerning the style and substance of pilot resource management.

These objectives grew out of earlier research which focused on

full-mission simulation scenarios in a Singer GAT-1 flight trainer and simple paper and pencil problem scenarios. The overriding consideration throughout all this research has been to apply human factors concepts to pilot information processing and decision making in order to:

- a) ascertain the role of pilot background, experience and knowledge in problem diagnosis and decision making; and
- b) describe the problem solving paths in sufficient detail to permit the ultimate development of various models of pilot behavior.

Major milestones in the total project development are noted in Figure 1. The particular tasks accomplished on the current research (CAT) are noted in Figure 2. The Aviation Psychology Symposium, listed as one of those tasks, provided the opportunity to exchange ideas with other researchers in the areas of pilot judgement and decision making. Computer Aided Testing Formats

Software for a touch sensitive CRT Computer Graphics Terminal has been successfully implemented to perform the biographical questionnaire, knowledge test and problem scenarios previously accomplished by paper and pencil testing. In addition, programs have been written which include dynamic navigation and control capabilities used as part of a full mission simulation embodying both problem diagnosis and destination diversion decisions. A prototype decision support system called Airplane Condition Evaluation (ACE) has also been developed as a computer aid for pilot decision making in emergency situations.

The programs are nearly experimenter free. Communication between subject and computer is accomplished by means of a touch sensitive CRT screen. When results of computer aided testing are compared with the equivalent paper and pencil data, the most striking difference concerns the number of inquiries and information tracks subjects employ.

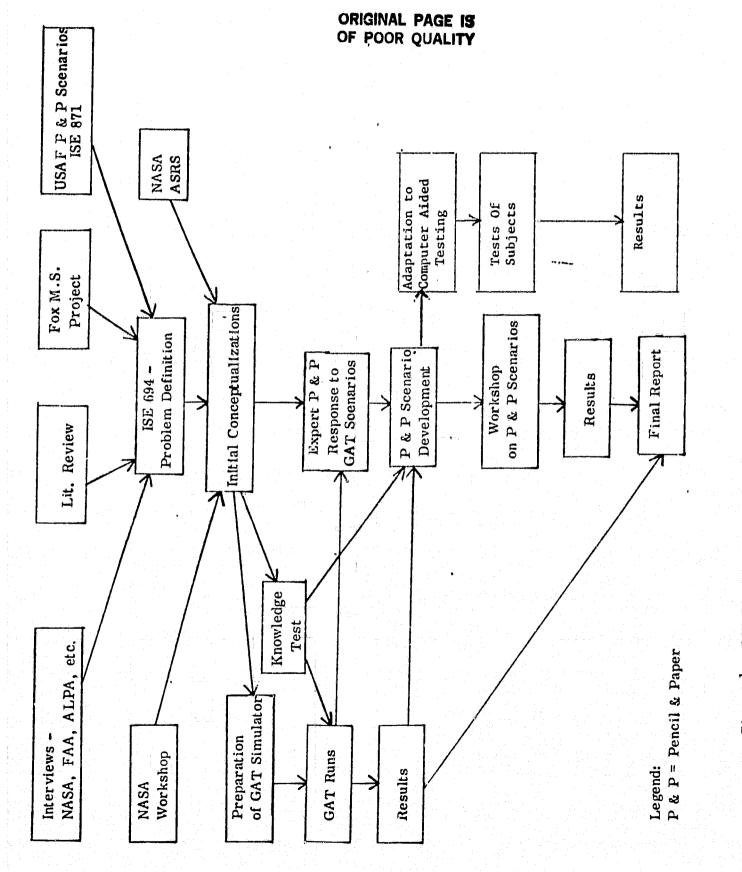


Figure 1. OSU ISE Department NASA Project Developments

OVERVIEW OF THE RESEARCH

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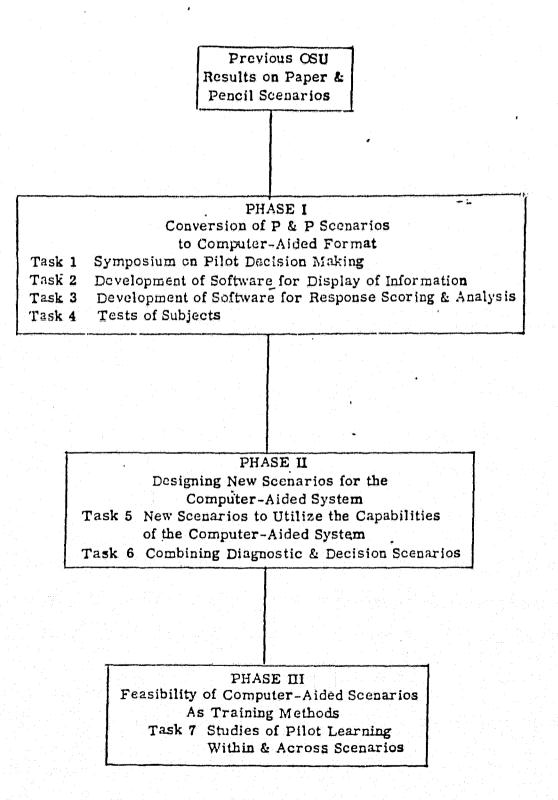


Figure 2: Overview of the Research

In most scenarios the paper and pencil subjects make fewer inquiries and employ fewer tracks than do the CAT subjects. The impersonal nature of communicating with the computer terminal as opposed to the verbal exchange in earlier paper and pencil studies appears to lessen inhibitions and broaden the alternative hypotheses considered by pilot subjects.

Results from CAT

As a result of analyzing the information seeking styles of some 40 subjects using computer aided testing, the following observations can be made:

- 1) More knowledgeable pilots make better diagnosticians than less knowledgeable ones.
- Pilots have difficulty in identifying the symptoms of a vacuum pump failure.
- 3) Knowledgeable pilots reach conclusion (right or wrong) more rapidly than others.
- 4) Less experienced pilots tend to use a larger number of diagnostic tracks than do more experienced pilots.
- 5) IFR rated airmen receive higher knowledge scores and higher diagnostic correctness scores than do VFR rated airmen.
- 6) High correctness scores are positively related to high mean time between inquiries.
- 7) Pilots follow a wide variety of different search patterns during problem diagnosis.
- 8) Individual pilots tend to exhibit similar search strategies across different diagnostic scenarios.
- 9) Efficient information searching can be recognized in particular scenarios and synthesized for general cases.
- 10) Problem solving schemata are scenario dependent.
- 11) There is no discernible relationship between the way a pilot collects information for diagnosis and the way he collects information for destination diversion decisions.

- 12) The only discernible learning effect across scenarios is that time between inquiries is reduced with subsequent trials.
- 13) Recorded performance measures depend more on the content of the scenario than on its order position within test session.
- 14) PLATO-GAT subjects exhibit resource management styles similar to those observed in full mission GAT simulations.
- 15) When faced with both problem diagnosis and the need for destination diversion decision making in the same scenario, pilots show a strong preoccupation with problem symptoms at the expense of positional awareness.
- 16) Pilots do not keep enroute alternatives in mind in case problems do develop. They react to emergencies more than they pre-plan for emergencies.
- 17) Data most often requested in diversion decisions are ceiling and visibility. Terrain receives a low number of inquiries.
- 18) Pilots often neglect available winds aloft information when selecting an alternate airport.

Potential

The computer aided testing instruments described in this report were developed as research tools to be used to better understand the decision making styles of pilots faced with critical in-flight events. However, based on repeated comments by subject pilots these tools may have even greater potential for pilot training. In addition to providing a wealth of potential simulated decision experiences, CAT could be used to uncover pilot deficiencies in their understanding of the nature of CIFE and to help them to develop more efficient search habits. This research can help pilots understand their own problem solving logic structure and hence enable them to reevaluate their approach to diagnosis.

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Chapter 1: Introduction

A. Background

Research concerning pilot and system response to critical in-flight events (CIFE) began in the OSU Department of Industrial and Systems Engineering with NASA contract NAS 2-10047 and was extended under NASA Grant NAG 2-75. Results of those efforts covering the period of October 27, 1978 to February 28, 1981 were reported in "An Investigation Into Pilot and System Response to Critical In-Flight Events" Final Report, Volume I and Volume II, June 1981 (6).

A critical in-flight event (CIFE) is defined as a situation that either develops quickly or over time which is unexpected, unplanned, and unanticipated and which is perceived by the pilot in command to threaten the safety of the aircraft. The CIFE is one which requires pilot judgment beyond routine decision making or preprogrammed decision structure. The safety of the aircraft depends more on pilot cognitive processes than skilled motor performance.

The overall objectives of the early research were to:

- Describe and define the scope of the critical in-flight event with emphasis on pilot management of available resources.
- (2) Develop detailed scenarios for both full mission and paper and pencil (P/P) testing of pilot response to CIFE's.
- (3) Develop statistical relationships among pilot characteristics and observed responses to CIFE's.

These objectives grew out of a concern with anomalies in reported accidents and incidents in which some pilots or crews seemed better able to handle unusual in-flight events than others.

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Initial efforts began with a concern for the dynamics of CIFE's and broad attempts to identify pertinent research issues. The final products were: (1) a set of scenarios with associated hardware and techniques for studying CIFE phenomena in a simple flight simulator; (2) a set of paper and pencil scenarios and associated techniques for studying pilot diagnostic strategies and diversion decision making processes; (3) a set of knowledge testing instruments designed to measure a pilot's understanding of aircraft subsystems and troubleshooting; (4) a study relating cockpit crew procedural compliance with performance errors. The result of these efforts were reported in the earlier study by Thomas H. Rockwell and Walter C. Giffin (6).

A five-phase model of pilot CIFE response was hypothesize i on the basis of (a) discussions with experts in industry and government and (b) observations made about pilot performance in both simulator and paper/pencil scenarios. The five phases were:

- (1) Detection
- (2) Diagnosis
- (3) Option generation
- (4) Decision making
- (5) Execution

Pilot information seeking activities permeated all five phases of this process.

GAT-1 Full Mission Simulation

All five phases of pilot response were studied in a GAT-1 flight trainer using the following three full mission scenarios: -2-

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- (1) Fuel starvation on the active tank (as might be encountered because of a loose fuel cap).
- (2) Partial power failure (as might be caused by a broken baffle in a muffler).
- (3) Navaid loss (as might be caused by failure of a single airborne receiver component).

A wide range of cockpit management styles and apparent skill levels were observed in these simulations. Although it was difficult to quantify, "good performance" was easily recognized by the observers of the experiment. The elements of "good performance" included:

- (1) professional use of the radio
- (2) precise heading and altitude control prior to and during the CIFE
- (3) constant awareness of the aircraft position along its intended route
- (4) prompt, but not instant, response to the onset of the CIFE (detection)
- (5) systematic procedure for troubleshooting
- (6) knowledge and use of available ATC resources
- (7) diversion decisions which allowed for further uncertainties.

In general, it was found that:

- Cockpit management style varies widely among pilots. For example, some are extremely self-reliant, others want immediate and extensive help from ATC while still others make the decision making process a joint effort with ATC.
- (2) Good stick and rudder pilots seem to have excess capability and maintain good stick and rudder performance ance during and after the CIFE. More marginal stick and rudder pilots, on the other hand, show increased frequency and amplitude of heading and altitude excursions, and experience communication difficulties in the face of a CIFE.
- (3) Pilots who score well on the knowledge test instruments tend to perform well in problem diagnosis and decision making.

Paper and Pencil Experiments

Paper and pencil (P/P) scenarios, and associated experimental techniques, were created to streamline the data collection and analysis for pilot responses to critical in-flight events. Although they lacked the high stress environment of the GAT-1 experiments, these scenarios did yield useful data on the pilot problem diagnosis and decision making strategy phases of pilot response to CIFE's.

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Four separate diagnostic problem situations were presented to forty subjects. These scenarios centered about problems presumed to be created by:

- (1) an oil leak at the oil-pressure gauge line
- (2) a vacuum pump failure
- (3) a right magneto drive gear failure
- (4) a frozen static port

The decision making phase of the paper and per_il experiments was an alternator failure during an IFR flight which forced a diversion decision on the pilot.

For the information seeking task required in the diversion decision, the pilot was supplied a simplified enroute chart with sixteen airports indicated by letter along his flight path. The subject was then given two minutes to ask for information about any of those airports. For each airport questioned, there were six pieces of information the experimenter was prepared to provide:

- (1) bearing and distance from his present location
- (2) ceiling at the airport
- (3) visibility at the airport
- (4) approach aids available

(5) ATC services available

(6) terrain surrounding the airport

The pilot's information seeking observations and ultimate airport selected were recorded.

Later, subjects ranked the suitability of these sixteen potential diversion airports. The worth, or weights, for the variables ATC, weather, time, and approach were obtained by regression analysis according to the techniques of conjoint measurement. This study was reported in Flathers, 1980 (1) and 1982 (2).

The following observations were made from the paper and pencil tests:

- (1) There is no correlation between knowledge score and total flight hours.
- (2) Knowledge score is correlated with pilot ratings held.
- (3) Pilots good in one section of the knowledge survey tend to be good in all sections.
- (4) Diagnosis performance is highly correlated with knowledge scores.
- (5) Knowledge is inversely related to total number of diagnostic inquiries, e.g., knowledgeable pilots reach conclusions (right or wrong) more rapidly than others.
- (6) Total diagnostic inquiries is inversely related to correctness, this implies that undirected experimentation is poor diagnosis style.
- (7) Total diagnosis correctness score is correlated with efficiency, i.e., the ability to arrive at a diagnosis with a minimum number of inquiries.
- (8) Civil trained pilots place a higher worth on ATC service in diversion decisions than do military pilots.
- (9) Private pilots place a higher worth on weather factors in diversion decisions than do commercial and ATP rated pilots.

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(10) ATP rated pilots place high worth on time in diversion decisions.

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- (11) Pilots with good diagnostic scores place less weight on approach aids in diversion decisions.
- (12) Pilots with good diagnostic scores place more weight on time in diversion decisions.
- (13) The pilots with good diagnostic performance were characterized as knowledgeable about aircraft systems, employed few tracks (a track represents a coherent line of questions, e.g., fuel systems), used few inquiries per track, and emphasized time in their destination diversion decisions.

B. Research Objectives

The major objective of the current research was to extend the effectiveness of past pilot diagnostic and decision testing through the use of computer interactive terminals employing existing software. Such interactive terminal systems provided the researcher with: (1) display graphics (e.g., segments of instrument panels with moving indicators), (2) real-time capability to allow timing of response elements, (3) rapid analysis of pilot inputs and (4) positive and/or negative feedback of results as a special experiment. Subject interactions with aircraft controls and displays were programmed as well as communication with ATC by selecting information requirements from an array ("menu") of information sources available.

The research plan entailed three major research phases embodying seven research tasks as shown in Figure 1-1. All of these tasks except Task 1 are detailed in the chapters to follow. Task 1, the symposium on pilot decision making, has been reported separately in Proceedings for the First International Symposium on Aviation Psychology (4). A table of contents for the proceedings is reproduced in Appendix G of Volume II of this report. -6-

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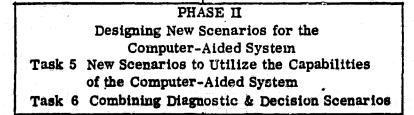
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OVERVIEW OF THE RESEARCH

Previous OSU Results on Paper & Pencil Scenarios PHASE I Conversion of P & P Scenarios to Computer-Aided Format Task 1 Symposium on Pilot Decision Making Task 2 Development of Software for Display of Information Task 3 Development of Software for Response Scoring & Analysis Task 4 Tests of Subjects



PHASE III Feasibility of Computer-Aided Scenarios As Training Methods Task 7 Studies of Pilot Learning Within & Across Scenarios Chapter II: Development of Computer Aided Formats

The major task in this research was to design and implement a computer aided testing device for studying pilot diagnosis and destination-diversion decision making. The initial software for the computer graphics terminal was designed around the problem scenarios, knowledge test, and biographical questionnaire used in previous paper and pencil studies (Tasks 2 and 3 of the project proposal). Some new diagnostic scenarios were also designed to take advantage of the capabilities of PLATO® (Task 5). In addition, both diagnostic and destination diversion scenarios were combined into a single scenario with added workload (Task 6).

A. Program Design

Appendices A to E in Volume II of this report include sample displays presented to the subject in the course of testing. The CIFE data collection/subject testing system was built using CDC's PLATO® system and the TUTOR programming language. Details on the various programming modules are discussed in Appendix F.

The goal was to produce a set of programs which could be nearly experimenter-free. Considerable effort was expended to provide the subject with detailed instructions via the graphics display on exactly what tasks were expected of him. After an initial sign-on via the terminal keyboard, all subsequent communication between subject and computer was accomplished by means of the touch sensitive CRT screen. The major elements of the program displayed for the subject included:

-8-

- (1) CIFE Router
- (2) Biographical Data
- (3) Knowledge Test
- (4) Six Diagnostic Scenarios
- (5) The Destination/Diversion Scenario
- (6) The Airport Ranking Exercise
- (7) VOR-Autopilot
- (8) Combined Destination Diversion and Diagnoszic Scenario Test (PGCIFE)
- (9) Data Display

Router

The CIFE Router is the main program module which allows access to all other modules and to the data files. The router is accessed by a special student sign-on to the PLATO® system. At the completion of each separate program module, control is returned to the Router where the subject may select the next module by touching the appropriate place on the CRT screen.

Biographical Data

The first time a subject uses the program he is assigned a unique subject identification number and is forced to enter the Biographical Data module in order to establish a subject data file. If testing is interrupted for any reason, the same subject number will permit access to any module without again going through the Biographical Data module. All biographical data are displayed in the form of multiple choice questions to which the subject responds by touch panel. A sample of biographical data questions is contained in Appendix A.

Knowledge Test

The Knowledge Test module presents a series of twenty multiple choice questions one at a time. These questions cover three subcategories: (1) Engine and Fuel Systems, (2) Electrical Systems and Cockpit Instrumentation, and (3) Weather and IFR Operations. The Knowledge Test displays are contained in Appendix A.

B. Diagnostic Scenario Operations

In each of the diagnostic scenarios the subject is told that he is flying a Piper Cherokee Arrow and is given a list of equipment and performance parameters. The next display is then a brief paragraph which describes his mission and the symptoms of a problem being encountered. The subject then has a fixed amount of time (usually four minutes) in which to seek information and arrive at a diagnosis of the problem. Information is available from four separate displays which can be called up by the subject at any time during his allotted test period. He is not penalized for the time required to paint a new display on the screen.

The information displays include: .

- (1) instrument panel
- (2) interior information
- (3) exterior information
- (4) ATC information

The instrument panel display contains most of the instruments and controls found in a Cherokee Arrow. Information is obtained by touching the appropriate dial or control. If, for example, a subject wants to know his oil pressure reading, he touches the oil pressure gauge, shown graphically on the panel layout, and its current status is printed for him in the display area. It is also possible to obtain information concerning control movement. For example, if the subject touches the prop rpm, a secondary input display permitting him to increase or decrease rpm appears. If he touches either of those boxes, an appropriate response message is then printed. The instrument panel display is shown in Appendix A.

The three remaining information displays all work alike. A subject touches the box containing desired information and a description of that information is printed out for him. Interior information refers to inside cabin conditions. This display permits one to find out about smoke, fluid leaks, unusual sounds, etc. which may be in the cockpit. Exterior information includes such things as cowling condition, wing condition, etc. A query on wing condition for example might bring forth a response "light rime ice visible". ATC information includes general weather and navigation aid status information of the type that a Flight Service Station or Air Route Traffic ©ontrol Center might provide. These data include such items as forecast winds aloft and freezing levels. All three displays, interior, exterior and ATC information, are shown in Appendix A.

When the subject reaches a conclusion concerning the cause of the problem (or he runs out of allotted time), he pushes the "give answer" choice available with each of the information displays. This action routes him to a second choice menu which permits him to seek further information in the event that he inadvertently or -11-

prematurely pushed "give answer". A second push of "give answer" then routes him to the lexicon illustrated in Appendix A. The subject formulates his diagnosis by touch panel entry designating up to nine of the words listed in the lexicon.

After entering his lexicon response the subject is asked a series of questions concerning his judgment of the problem. The computer asks for an estimate of how long the plane will fly with its existing problem, how critical (scale 1 to 7) the problem is, and how confident the subject is in his own diagnosis.

At that point the problem around which the scenario was designed is presented to the subject as the correct diagnosis. He then is again asked to respond to the time and criticality questions in light of his now complete knowledge of the state of his aircraft systems. Control than passes back to the router where a new scenario or other task can be selected.

A complete time history of all subject data inquiries is maintained for each individual subject tested. The particular display and the item on (at display which was queried are noted together with the time since initiation of that scenario at which the query took place. This time history of information search is available through the Data Display module.

C. Diagnostic Scenario Content

Diagnostic scenarios one through four are adaptations of scenarios previously used in paper and pencil testing. (See project report NAS2-10047 (6)). They concern: (1) an oil pressure gauge line break, (2) a vacuum pump failure, (3) a broken magneto drive gear, and (4) a blocked static port. Two new scenarios were created to accomplish certain computer aided testing objectives which were not part of the earlier paper and pencil tests. Scenario five is a nearly no-win situation. The problem is manifested by a partial power loss which no amount of cockpit experimentation can correct or even totally identify. The power loss is assumed to be caused by a broken baffle in the muffler which creates a horsepower robbing backpressure in the exhaust system. This scenario is designed to explore a subject's information search patterns while eliminating the possibility of accidently uncovering the key element of information which uniquely identifies each of the four earlier scenarios. Scenario five also forms the diagnostic portion of the combined destination diversion and diagnostic search task to be described later.

Scenario six is designed to present symptoms which might be improperly attributed to a mechanical failure when in fact a nonmechanical act has caused the problem. This scenario concerns an upholstery fire caused by a carelessly discarded cigarette.

The complete text of each scenario and the proper diagnosis as presented to the subject are reproduced below and in Appendix B. -13-

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Diagnostic Scenario One

You are making a day trip from Albany,NY to Burlington,VT. You fly out of Albany at 9:00am, cleared Victor-91, Burlington. You climb to a cruising altitude of 7000ft. After 20 minutes of routine IMC flying you notice the smell of engine oil.

How would you diagnose the problem?

Our diagnosis of the problem was the following:

A small crack developed in the oil line feeding the oil pressure gauge. This crack reduced the oil pressure reading drastically, but did not seriously affect the actual lubrication of the engine. A small pool of oil began to form on the floor of the cabin, pilot's side. Assuming that the cracked line would not deteriorate quickly into a complete break, you were in no immediate danger of engine seizure.

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Diagnostic Scenario 2

You are making a day trip from Augusta,ME to Lebanon,NH.You fly out of Augusta at 9:00 am, cleared Victor 39 to Neets intersection, Victor 496 to Lebanon.You climb to a cruising altitude of 6000 ft.After 15 minutes of routine IMC flying in instrument conditions, your instruments indicate an increase in airspeed and steadily decreasing altitude while maintaining level flight attitude.

How would you diagnose the problem?

Our diagnosis of the problem was the following:

Your vacuum pump failed as indicated by the low reading of the suction gauge. The vacuum pump drives the attitude and directional gyros. As the artificial horizon lost its drive it started to sag to the right and you compensated by turning left, leveling the artificial horizon and putting the plane in a slow, descending left bank. The airspeed increase was due to the slight nose-down attitude.

Diagnostic Scenario 3

You are making a day trip from Keene,NH to Montpelier,VT. You fly out of Keene at 10:30 am, cleared Victor-151 to Montpelier. You climb to a cruising altitude of 5000 ft. After 20 minutes of routine cruise in IMC your engine suddenly starts running extremely rough, shaking the whole plane and losing about 20% of its cruise power.

How would you diagnose the problem?

Our diagnosis of the problem was the following:

Your engine suffered a broken drive gear in the right magneto. The resultant untimed ignition conflicted with the remaining good ignition and caused the extremely rough engine and backfiring. Switching from 'both' to the left magneto would have resulted in a smooth running engine with slightly less power than normal cruise. Ŷ!

Diagnostic Scenario 4

You are making a day trip from Sanford, ME to Messena, NY. You fly out of Sanford at 8:30am, cleared Victor-496 to Lebanon, Victor-141 to Messena You climb to a cruise altitude of 6000. After 20 min IMC flying, Boston Center instructs you to climb and maintain 10,000ft.You acknowledge and begin your climb between layers.After 2 min of climb, you notice your indicated airspeed dropping off steadily from 100kts, maintaining constant pitch attitude.

How would you diagnose the problem?

Our diagnosis of the problem was the following:

As you climbed through 6500ft, the static port froze over as the outside air temperature dropped below 32°F. This caused the airspeed indicator to decrease as altitude increased and the VSI and altimeter to read low. Several connective actions were possible: return to your previous altitude of 6000ft; open the alternate static source; break the VSI glass.

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Diagnostic Scenario 5

You are making a day trip from Augusta, ME to Lebanon, NH. You fly out of Augusta at 10mm, cleared Victor 39 to Neets Intersection, Victor 496 to Lebanon. You climb to a crusing altitude of 8000ft. After 20 minutes of routine flying in instrument conditions with light to moderate turbulence, you notice that increased nose-up trim is required to maintain a constant indicated altitude and that your IAS has decreased 20kts from normal cruise.

How would you diagnose the problem?

Our diagnosis of the problem was the following:

A baffle was broken in the muffler. The broken baffle partially blocked the exhaust system causing increased exhaust backpressure. The increased exhaust backpressure absorbed a portion of the available horsepower output from the engine. With a constant throttle setting, the prop flattened pitch to maintain constant RPM causing a decrease in airspeed while altitude was held constant. Conversely when airspeed was held constant altitude decreased due to reduced power output available at the propeller. Increasing the manifold pressure with added throttle permitted enough power to be developed to maintain altitude at a greatly reduced airspeed.

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Diagnostic Scenario 6

yeu are making a day trip from Montpelier.VT to Bangor.ME with two passengers on board. You fly out of Montpelier at 1:00pm, cleared radar vectors to Wylie intersection, direct Augusta, Victor 3 to Bangor. You climb to a cruising altitude of 9000ft. After 30 minutes of routine flying in instrument conditions with light to moderate turbulence, one of your passengers reports smelling a faint burning odor. You are unable to detect the odor because you have a head cold.

What is the first thing you would do?

Our diagnosis of the problem was the following:

Rear seat carpeting was smoldering. The rear seat passenger lit a cigarette shortly after takeoff. When he disposed of it in the ashtray, it was not completely extinguished. The cigarette fell down from the ashtray and was beginning to char upholstery material. The fire was easily extinguished, once recognized and posed no immediate danger to the flight. -19-

D. The Destination Diversion Scenario

The destination diversion scenario was also an adaptation of an earlier scenario used in paper and pencil tests. The goal was to examine the information seeking activities and decision making strategy of subjects forced to make an enroute diversion decision.

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The subject is presented with a business trip scenario involving an IFR flight from Bangor, Maine to Glens Falls, New York in a Piper Cherokee Arrow. Area weather charts, a Flight Service Station briefing and a flight plan are presented in turn as the scenario begins. Once the flight is underway, a simplified low altitude enroute chart is shown on the graphics display. A flashing arrow on that chart depicts the progress of the flight as the scenario continues to depict flight progress along the intended path. At a point midway along the planned flight the subject is told that he has experienced an alternator failure and must now operate on battery power alone. The maximum expected life of the battery is insufficient to carry him on to his planned destination. The problem now is to select a suitable alternate airport.

Alternate airports are depicted by number on a further simplified enroute chart. The subject may ask for up to six items of information about any of the sixteen airports depicted on the chart:

- (1) bearing and distance
- (2) ceiling
- (3) visibility
- (4) approach aids

-20-

(5) ATC services

(6) terrain

He has two minutes in which to conduct his information search and select an alternate airport. A summary display of all information requested is available to the subject anytime he wishes to examine it.

A complete time history of his information search is maintained in the computer data file along with his final choice of airport. A complete set of graphics displays shown to the subject during testing is shown in Appendix D. A sample of the stored data is contained in Appendix C.

Airport Ranking

As an alternative way of studying the diversion decision, an airport ranking exercise was created. The ranking exercise was the alternator failure scenario previously discussed. This module accomplishes the same thing by way of computer graphics that was previously accomplished by shuffling cards in the paper and pencil studies.

Here the subject is presented with a matrix of information involving sixteen possible diversion airports. Each airport is listed together with its ATC services, ceiling, visibility, time to reach and approach aids. The subject ranks these airports from most desirable to least desirable. All entries are made by touch panel. The matrix is updated with each ranking decision by interchanging rows, replacing the current occupant in decision row x with the occupant the subject would prefer while moving the old occupant in x to the position vacated by the new choice. The displays used in this module are shown in Appendix D.

E. VOR-Autopilot

The VOR-autopilot module is a training module designed to instruct a subject on the use of the dynamic navigation and control capabilities he will later use in the combined destination diversion and diagnostic scenarios. The basic control display is shown in Appendix E.

By appropriate touch panel entry a subject can select desired heading and altitude commands for his aircraft autopilot. If the command heading is not the one currently displayed on the directional gyro, the simulated aircraft will begin a standard rate turn toward that heading. If the altitude selected is not that currently stored in the instrument panel display, the simulated aircraft will begin a pre-programmed rate of climb (or descent). At the same time that instrument indications are being updated, the simulated aircraft continues to move through space. That movement is reflected by the tunable VOR heads contained on the display.

The subject can determine his current position in space by selecting appropriate VOR frequencies and centering the needles. The admissable VOR's are depicted on an accompanying simplified low altitude chart.

Each subject is given a few brief training exercises involving turns to heading and locating the aircraft relative to VOR stations. This training exercise does not refer to any particular problem scenario but rather is intended to familiarize him with the new dynamic system.

F. Combining Destination Diversion With Diagnosis

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The combined scenario described in Figure 2-1 was designed to meet two purposes:

- (1) It was a "no win" problem, i.e. no pilot would really be able to find the true cause from the symptoms presented. Hence, it would avoid a pilot stumbling into the key information item and force the pilot to examine all possible hypotheses. The diagnosis portion was the same problem previously discussed as scenario #5.
- (2) It combined on the touch CRT both the destination diversion decision and the problem of diagnosis.

Appendix E shows the displays used in this scenario. This includes a simulated low altitude chart. A program has been developed for PLATO® which locates the aircraft relative to the VOR site, allows heading changes and allows the position of the aircraft to be determined from VOR radials as the aircraft moves at some selected speed and heading. It should be noted that attempts to develop positional awareness are used throughout. These include position reports to ATC based on dual VOR's location assessment, auto pilot heading change requirements, new clearances and even a concerned passenger requesting a return to the departure airport. This scenario employs some dynamics, i.e. loss of altitude with neutral forces on the yoke and attendant changes in VSI and airspeed with control inputs. As the flight progresses the VOR needle shows a deflection. The scenario permits communication with ATC and allows for declaring emergencies. -23-

Consult attached simplified altitude chart.

You are on an IFR flight from Utah Municipal Airport to Haven Count Airport. You depart on V-110 at 6000 ft. in your Cherokee Arrow (N123B) which is equipped with a 3-axis autopilot. There is a NOTAM out which reports that Colorado VOR is out of service during the period you plan to navigate. <u>Navigate using Ohigh and California VORs</u>. You have been enroute 60 minutes from Utah Municipal Airport. You are on the gauges but the ride is smooth. Weather briefing indicated that winds at 6000 were expected to be light and variable.

You have one passenger aboard.

Weather at:

Haven County Airport = 2000 & 5 Ohigh = 1000 & 3 Wind Falls = 1000 & 3 by a C-172 (10 Minutes Ago)

Cleve Center calls and reports radar contact is lost. Please report present condition.

Clearance

ATC Response:

N123B, thanks for the position report. Here is your new clearance: proceed direct California VOR direct Haven County Airport at 6000.

There will be opposite traffic at 5000 . . . maintain 6000.

Please confirm your new heading and altitude after your turn.

Scenario Change

While practicing hand flying with your autopilot disengaged, you notice that increased nose-up trim is required to maintain a constant indicated altitude and that your IAS has decreased 20 kts. from normal cruise.

Your passenger notes this problem, and suggests that you turn back to Utah Municipal.

Determine the nature of the problem, and your destination decision.

Figure 2-1: Combined Diagnosis and Destination Diversion Scenario

The purpose in combining diagnosis and destination diversion problems into a single scenario was outlined in Task 6 of the original proposal. The combined scenario was created to provide a more realistic framework for tests which would begin to approach the fidelity of the earlier GAT simulations, in the sense of dynamic position information and added workload. The research question was whether PLATO[®] lends itself well to simultaneous navigation and diagnosis even though no control manipulation skills beyond selection of heading and altitude commands were possible. -25-

Chapter III: Results of Computer Aided Testing

Two types of analyses were made based on the data generated through computer aided testing. The first type of analysis was based on the power of the computer to generate a variety of summary statistics. Here such standard techniques as regression analysis, t-tests, chi-square tests and frequency counts were used on a variety of different combinations of subject data. The intent was to isolate important performance measures and classify groups of subjects.

The second type of analysis required the generation of creative graphical aids which permit a particular subject's information seeking strategy to be absorbed at a glance. By comparing performance across subjects it became possible to identify several distinct search strategies which were not apparent from the more formal statistical tests. The graphical aids used to depict information seeking behavior during diagnosis testing were the pilot information plot (PIP) and schema diagrams. The aid used to depict the destination diversion information search was the destination information graph (DIG).

A. <u>Depiction of Diagnostic Information Seeking Patterns For</u> Individual Subjects

Figures 3-2 through 3-5 depict a way to view subject information seeking patterns during diagnosis testing. Sources within logic tracks are identified for each scenario. The pilot information plots (PIP's) are a quick way to visualize:

- (a) the number of tracks * employed
- (b) the order of inquiries within and between tracks
- (c) the time between inquiries
- (d) the number of track returns and the information resampled

Using these PIP's various information seeking strategies can be observed. For the suction failure problem as shown in Figure 3-1, Figure 3-2 depicts a subject with a logical and efficient approach to diagnosis. Figure 3-3 depicts an almost random inquiry which leade to no logical conclusion. For most scenarios there is a key element or piece of information to identify the problem, e.g. low suction and vacuum failure. Figure 3-4 shows a subject with the key piece of information but who still does not recognize the correct answer. Figure 3-5 clearly indicates a subject using a systematic approach but an incorrect one. In this case the subject believes ice to be the causal factor for the symptoms presented.

Based on PIP analyses, idealized information searching can be hypothesized. The ideal pilot first confirms the symptoms given him. He then establishes whether his engine status is threatened by whatever cause lies behind the symptoms. Usually oil temperature and pressure and manifold pressure suffice to test this condition. Next he makes two or more hypotheses as to the cause, and makes a determination of the plausibility of these hypotheses with a minimal number of inquiries within the appropriate tracks. He rarely needs to go over old logic tracks sampled. Finally, given a logical cause of the symptoms (usually from the key information element), he will often make sure alternative hypotheses are still not viable by additional information inquiries.

*a track is a coherent line of questioning focused on an aircraft subsystem, e.g., internal engine condition.

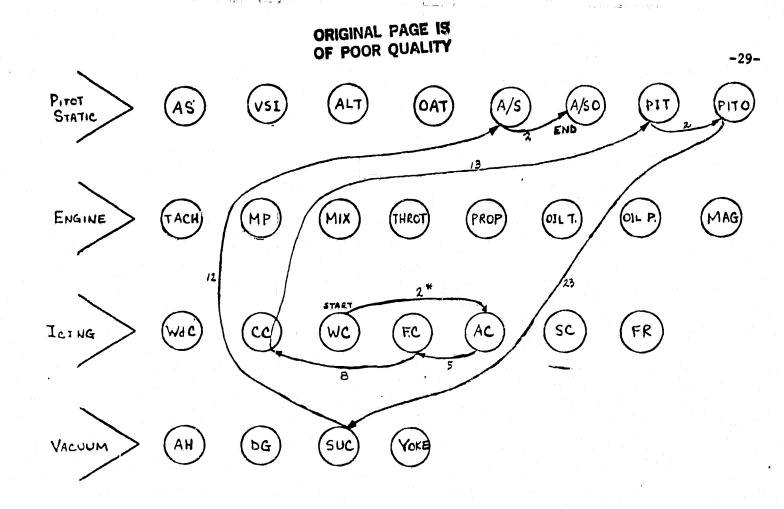
SCENARIO

You are making a day trip from Augusta, ME to Lebanon, NH. You fly out of Augusta at 9:00 a.m., cleared Victor 39 to Neets intersection, Victor 496 to Lebanon. You climb to a cruising altitude of 6000 ft. After 15 minutes of routine IMC flying in instrument conditions, your instruments indicate an increase in airspeed and steadily decreasing altitude while maintaining level flight attitude. How would you identify your problem?

Our Diagnosis of the Problem was the Following:

Your vacuum pump failed as indicated by the low reading of the suction gauge. The vacuum pump drives the attitude and directional gyros. As the artificial horizon lost its drive it started to sag to the right and you compensated by turning left, leveling the artificial horizon and putting the plane in a slow, descending left bank. The airspeed increase was due to the slight nose-down attitude.

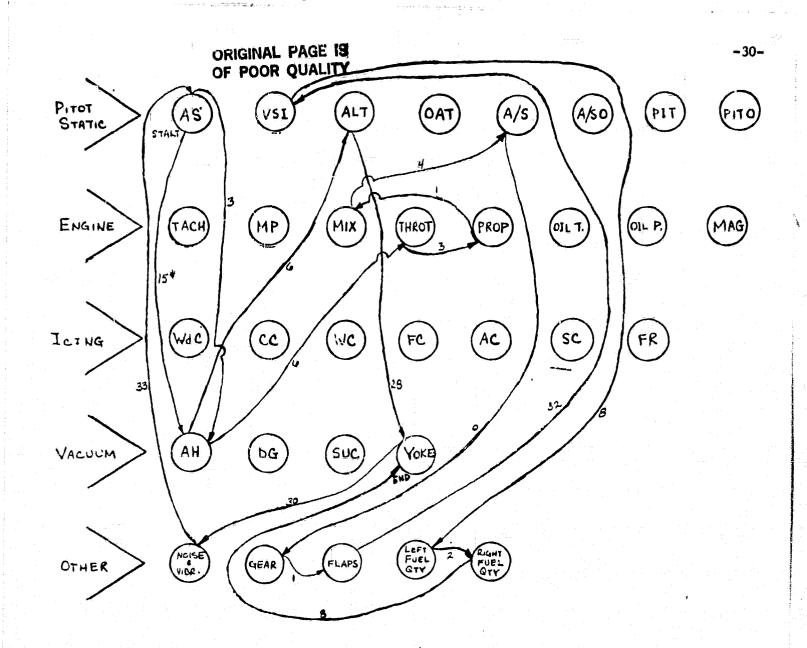
Figure 3-1: Suction Failure Problem and Diagnosis



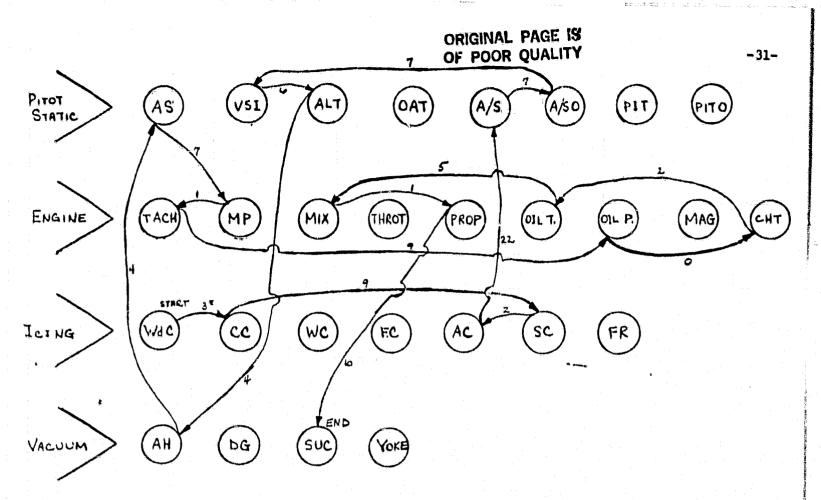
OTHER

RESPONSE: Gyro Broken CONFIDENCE IN RESPONSE: 7 CRITICALITY: 5 TOTAL FLIGHT TIME: 100-300 Hrs. TOTAL DIAGNOSIS TIME: 78 Sec. RATING: Private - VFR KEY TERM: NY TOTAL NO. INQUIRIES: 9 KNOWLEDGE SCORE: 60% *Time Between Inquiries

Figure 3-2. Pilot Information Plot For Scenario #2 Subject #53



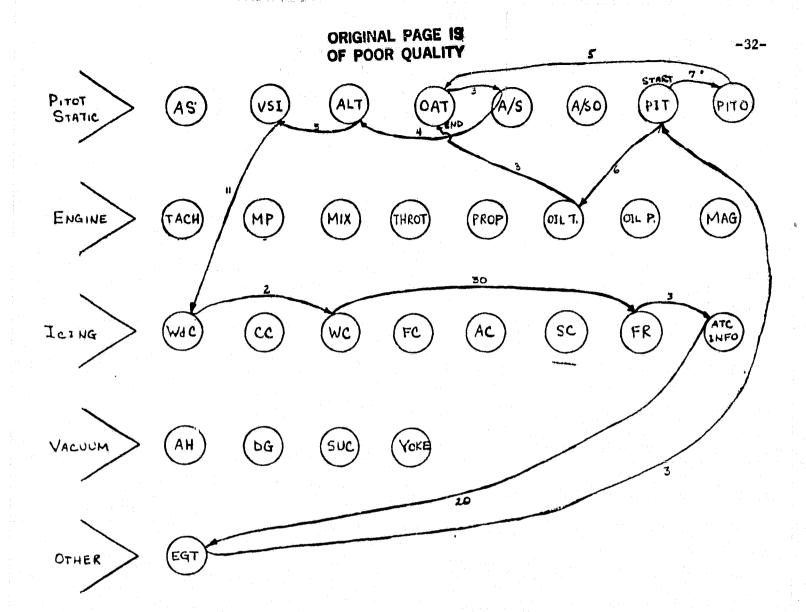
RESPONSE: Elevator Control Pressure CONFIDENCE IN RESPONSE: 7 CRITICALITY: 1 TOTAL FLIGHT TIME: 100 Hrs. TOTAL DIAGNOSIS TIME: 211 Sec. RATING: Private - VFR KEY TERM: N Y TOTAL NO. INQUIRIES: 20 KNOWLEDGE SCORE: 50% *Time Between Inquiries



OTHER

RESPONSE: Pitot Ice CONFIDENCE IN RESPONSE: 6 CRITICALITY: 3 TOTAL FLIGHT TIME: 501-1000 Hrs. TOTAL DIAGNOSIS TIME: 127 Sec. RATING: Commercial-IFR KEY TERM: N Y TOTAL NO. INQUIRIES: 19 KNOWLEDGE SCORE: 50% *Time Between Inquiries

Figure 3-4. Pilot Information Plot For Scenario #2 Subject #55



RESPONSE: Pitot Static Prop Ice CONFIDENCE IN RESPONSE: 6 CRITICALITY: 6 TOTAL FLIGHT TIME: 501-1000 Hrs. TOTAL DIAGNUSIS TIME: 123 Sec. RATING: Commercial-IFR KEY TERM: N Y TOTAL NO. INQUIRIES: 16 KNOWLEDGE SCORE: 40% *Time Between Inquiries

Figure 3-5. Pilot Information Plot For Scenario #2 Subject #54

B. Summarizing Diagnostic Information Search Patterns Graphically

Schema theory provides a method for investigating the underlying logical structure (schema) which provides a mental outline by which a pilot may organize information. A pilot has some knowledge of the events that typically occur in a critical in-flight event and the order in which they take place. The completeness of his knowledge will determine the ease with which he can comprehend and summarize diagnostic information. By separating "correct" diagnoses from "incorrect" diagnoses one may be able to develop guidelines from schemata which highlight efficient information search strategies.

The schema diagrams developed for this research plot each item of information sought by the frequency with which it is requested and the median order percentile in which that item appears in the total stream of information requests. The technique is patterned after that discussed by Geiselman and Samet (3).

For example, on scenario 1 subject 64 asked for 26 separate items of information over a period of 205 seconds. Cylinder head temperature was the third item of information sought and that was requested at 8 seconds into his search. In terms of order, this information was in the 11.5 percentile order position and the 3.9 percentile time position. By performing similar calculations for each subject it was determined that 90 percent of the subjects in scenario 1 with the correct diagnosis asked for cylinder head temperature (CHT), that the median order percentile for their requests was 34.5. These data were then plotted on two schemata, one showing frequency versus order and the second showing frequency versus time.

By grouping items of information into common subject matter clusters, it is possible to develop a two-dimensional hierarchical outline of subordination and sequential order of information. Level of subordination is measured by the percent of pilots who include that information in their search. Output position is standardized by computing a median output position percentile for each item of information which was sought by at least one subject.

Figure 3-6 shows a schema diagram for successful subjects on the first scenario. This scenario involved a cracked oil pressure gauge line behind the instrument panel. Positive confirmation of the problem was possible by noting the reduced oil pressure and the presence of fluid leaks in the cockpit. Roughly 90 percent of the subjects included oil pressure, oil temperature, cylinder head temperature and fluid leaks in their search, although fluid leaks did show up late in the order of information (cluster 1 on the schema diagram). Another sizeable portion, 60 percent, appear to be searching for evidence of an oil system failure in front of the firewall (cluster 2). A much smaller minority of approximately 10 percent appear to be concerned about general engine health (cluster 3). The information in cluster 4 which appears late in the search process for some of the subjects seems to be directed toward localizing the source of the smell of hot engine oil, e.g. is it in or outside of the cabin. Note that even though all of the

Frequency (%) vs. Order (Median)

Subjects: 42, 43, 46, 47, 51, 52, 53, 54, 55, 56, 57, 59, 60, 61, 64, 67, 68, 80

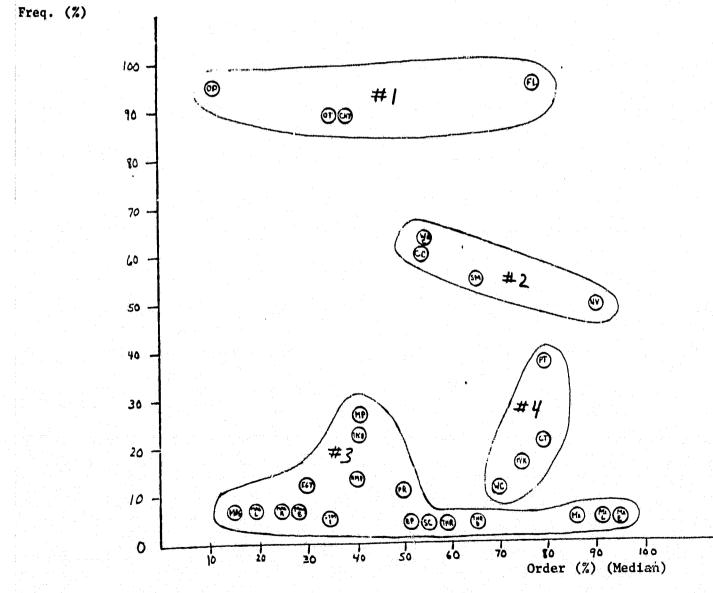


Figure 3-6: Schema For Subjects In Scenario #1 With Correct Diagnosis (See Figure 3-21 for the list of symbols used in schemata on Page 89)

-35-

subjects in this schema solved the problem, the median percentile position for the key item "fluid leaks" was only 78. Evidently many sought further confirmation even after having the key. As identified earlier this is an example of pilots checking on alternative hypotheses.

C. Grading Diagnostic Results

Correctness scores, which were used to measure a subject's diagnostic performance, were related to a simple five point scale. Each of the four basic scenarios could have been unambiguously diagnosed if the subject asked for a certain key piece of information. For example, in the oil pressure gauge line leak scenario the key element was "fluid leaks". In the vacuum pump failure scenario the key element was "suction". In the magneto gear failure scenario the key element was "left mag". In the static port icing scenario the key element was "alternate static source". Correct answers were awarded a grade of five, four, or one, depending upon whether the key element was asked for, whether the subject was on the right track or whether there was no apparent connection between the information sought and the correct response.

Incorrect answers which could at least be partially supported from a logical search pattern were given two points. Incorrect and illogical searches were graded zero. This scheme is summarized in the table below.

| Conclusion Information | Correct Answer | Incorrect Answer | Incorrect and Illogical |
|---------------------------|-------------------|---------------------|----------------------------|
| Key | 5 | 2 | 0 |
| No Key But Right Track | 4 | 2 | 0 |
| Never on Track | 1 | 2 | 0 |

Table 3-1. Diagnostic Scoring

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-36~

The relative grade frequencies observed by scenario were

as follows:

a programme de la segur

Table 3.2

| C | onclusion | Correct Answer % of all subjects | Incorrect But Logical | Incorrect and Illogical |
|---------------------------|----------------------------------------------------------|-------------------------------------------|--------------------------------|--------------------------------|
| Кеу | Scenario #1 Scenario #2 Scenario #3 Scenario #4 | 52.4% 23.8% 63.6% 43.2% | 0% 2.4% 0% 2.7% | 9.5% 4.8% 3.0% 0% |
| No Key But Right Track | Scenario #1 Scenario #2 Scenario #3 Scenario #4 | 4.8% 2.4% 3.0% 2.7% | 19.0% 2.4% 9.1% 29.7% | 14.3% 4.8% 6.1% 13.5% |
| Not on Track | Scenario #1 Scenario #2 Scenario #3 Scenario #4 | 0% 0% 0% 0% | 0% 4.8% 0% 8.1% | 0% 54.8% 15.2% 0% |

Observed Relative Frequency By Scenario

From these data it would appear that scenario 3 was the least ambiguous of all. Nearly 65 percent of the subjects taking the mag failure scenario found the key and correctly identified the problem. Scenario 2, the vacuum pump failure appears to be the most difficult. Nearly 55 percent of the subjects taking that scenario drew conclusions which could not be supported by the data they collected. In most cases they seemed to have an ice fixation and blamed their symptoms on a frozen pitot-static system in spite of no collaborating evidence. When faced with a genuine frozen static port scenario they were much more logical with only

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13 percent drawing conclusions which could not be supported by data. This may reflect the fact that many of these subjects seemed obsessed with ice problems and blamed ice for any conflicting flight instrument readings.

D. Depiction of Destination Diversion Information Seeking Patterns

Appendix D presents the CRT displays given to the subject for the destination diversion phase of testing. In this scenario the pilot loses his alternator in IMC conditions beyond the range of his destination or departure airports. He must choose between alternate airports with different attributes. Such attributes include ceiling, visibility, bearing and distance, navigation aids, presence of ATC support and terrain. In this test there is no absolute answer; rather, the test seeks to see how pilots weigh information about alternative airports.

The Destination Information Graph (DJG) provides a quick way to summarize the information seeking style of an individual subject. The DIG's portray:

- (a) the type of information sought
- (b) the frequencies with which information items are queried
- (c) the order in which airports are considered
- (d) the ultimate choice of which alternate airport is selected and how much data that decision required compared to airports considered but not selected

Figure 3-7 illustrates a typical DIG. This particular subject asked for information on only three airports, although his search was nearly total over the information available for each. The airport selected, number 6, was the first one he searched.

1.1. BORA 42.00

Subject #67 Time Lapsed: 134 Sec./Port Selection: 13.33 Sec./Inquiry: 5.875 Airport Selected: #6

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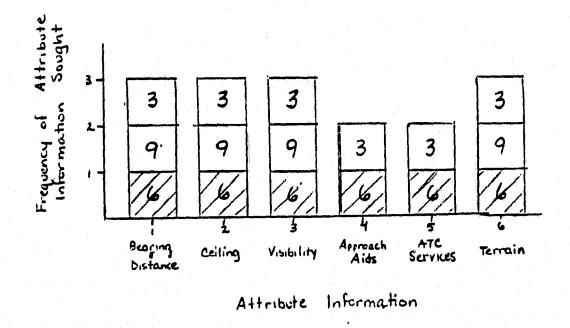


Figure 3-7. Destination-Diversion Quick Decision

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E. Subject Background

A total of forty-two subjects participated in the computer aided testing (CAT) program. All were rated pilots ranging from Private to ATP. Four subjects were used for special purpose testing leaving thirty-eight who participated in full sets of CAT experiments. Forty-two percent were private pilots, fortytwo percent were commercial pilots and sixteen percent held the ATP rating. Eighty-eight percent of the private pilots were not instrument rated. This group was recruited to investigate whether or not the CAT scenarios could separate performance of pilots with IFR versus VFR experience.

Total flying time among subjects ranged from fifty hours to 15,000 hours, the average being 1,654 hours. Their single-engine experience ranged from fifty to 7,500 with an average of 933 hours. Instrument flying experience, which in the case of non-rated private pilots included elementary hood time, averaged 257 hours. The most experienced instrument pilot reported 1,500 hours of instrument flying experience. Histograms reflecting this broad range of flying experience among subjects are shown in Figures 3-8, 3-9, and 3-10.

Knowledge survey scores ranged from twenty-five to eighty percent with an average of 56.6. Although individual subjects were weaker in some areas than others, the average percent correct by category across all subjects was nearly constant. The distribution of knowledge scores is given by Figure 3-11.

A brief summary of pertinent biographical information for each subject is given in Table 3-3. A complete subject data record,

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Figure 3-8: Distribution of Total Flying Experience

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Figure 3-10: Distribution of Instrument Flying Experience

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Figure 3-11: Distribution of Knowledge Scores

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including response to all scenarios is available in the Master Data Table, Table 3-5. A description of the variables contained in the Master Data Table is available in Table 3-4.

F. Diagnostic Performance

Means and standard deviations for all performance variables are listed in Table 3-6. Correctness scores are portrayed in Figure 3-12. This histogram illustrates an extremely wide spread in performance. Nearly eight percent of the subjects completely missed every scenario for a net score of zero while only three percent were able to achieve a perfect correctness score. The average percent total correct was 47. Individual scenario scores are more enlightening. Out of a maximum of five points for each scenario, the averages are 3.31, 1.46, 3.33, and 3.00 for scenarios one through four respectively. It is obvious that scenario two, the vacuum pump failure, posed significantly more problems in diagnosis than did the others. This is especially noteworthy in view of the current failure rates in general aviation vacuum pumps. These data seem to indicate that pilots may have difficulty in recognizing the symptoms of such failures.

Regression Analysis

In order to investigate what factors predict performance measures, a series of stepwise regressions were performed. These are listed in Table 3-7 and 3-8. Each table indicates the dependent variable, R^2 , N, and three sets of independent variables (biographical and experience, knowledge, independent performance variables) from which the predictor variables are chosen. Table 3-7 indicates the regression models attained for the correctness scores on scenarios one through four. Predictive variables included knowledge survey score, number of inquiries, number of tracks, and the mean delta (decision and response) time. Maximum R-square improvement was chosen. A significance level (α) of 0.1 was chosen. Four subjects' (69-73) time data had to be deleted since they were given unlimited time to complete a protocol analysis. The MAXR stepwise technique employed by SAS will eliminate a subject's data from consideration if that subject has any missing values for the variables included in the regression procedure. This accounts for the removal of seventeen subjects from the first set of regression models shown in Table 3-7.

The general strategy was to include all predictor (independent) variables, even those which could be related to each other, e.g. DIFTT (total tracks - total unique tracks) and total tracks and total unique tracks. However, no variable could be introduced if it had been derived from another significant predictor variable. On the other hand, no predictor variable which formed part of the dependent variable was allowed in the regression. For example, since TOTCORR = Cl + C2 + C3 + C4, a predictor variable for TOTCORR could not be introduced which contained either Cl, C2, C3, or C4, e.g., Z_t . $(Z_t = total correct/total tracks)$. The reader is referred to Table 3-4 for complete definitions of symbols and terms.

The strategy employed in deriving the regression models presented in Table 3-8 is slightly different from that described above. The objective here was to condense the candidate predictor -46-

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Figure 3-12: Distribution of Diagnostic Scores

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Table 3-3. Subject Biographical Information

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KNONLEDGE SURVEY

PERCENTAGE SCORE ON KNOWLEDGE SURVEY CATEGORY SCORE ON KNOWLEDGE SURVEY 1=KNOWLENGE SUBSCORE FOR ENGINE AND FUEL SYSTEMS 2=KNOWLEDGE SUBSCORE FOR ELECTRICAL SYSTEMS AND COCKPIT INST. 3=KNOWLEDGE SUBSCORE FOR WEATHER AND IFR OPENATIONS DESTINATION DIVERSION TEST 3=VISTBILITY 4=APPROACH AIDS 5=ATC SERVICES AVAILABLE 6='TERRAIN SCORE ORDER PORT INFO

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Table 3-5. Master Data Table

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| | 2 2<24 | 13.7692 | 6.000 | 4.4375 | 5.7778 | 6.7813 | 13.2941 | 7.6000 | 13, 8000 | 3.8750 | | 2.26987 | 2.4000 | • | 2.73333 | 3.0000 | 3.41176 | 5.00000 | 2.85714 | 2.42857 | |
|----|---------------|------------|-------------------|------------|------------|-----------|-----------|--------------|-----------------------|-----------|--------------------|------------------------|---------------|--------|-----------|-----------|-----------|-----------|-----------|-----------|--|
| | ><#8 | 213.721 | 29.133 | • | 49.211 | 96.418 | 55.455 | 54.942 | 55.810 | 16.712 | | •. 163 •4 3 · 3 | 0.200000 | • | 0.041667 | 0.187500 | 0.102941 | 0.096154 | 0.357143 | 0.100889 | |
| | en «es | 13.7059 | 6.6250 | ٠ | 7.5888 | 9.5333 | 8-9999 | 6.8571 | 7.1429 | 3.7073 | <>0404-E | 13.00 | 12.00 | 21,00 | 20.50 | 15.00 | 14.50 | 16.25 | 5.00 | 21.25 | |
| | ><#4 | 241.00 | 224-76 | 270.65 | 153.43 | 462.57 | 313.17 | 156.03 | 12.33 | 22.10 | <>0004-z | 5 4.25 | 5 4.50 | • | | 6 3.50 | 0 3.50 | 0 2.50 | 0 3.50 | 9 4.75 | |
| | ¥2<×0 | 18. 0000 | 12.2632 | 10.6190 | 9-0000 | 17.2857 | 18.9167 | 11.2308 | 5.3333 | 5.5926 | <>UDE-OF | 4.25 1.75 | 3.25 1.75 | • | 3.59 2.00 | 3.50 1.50 | 3.00 1.00 | 2.50 1.00 | 1.75 1.00 | 4.00 2.50 | |
| | > 4 H - | 337.778 18 | 93. 000 12 | 34.489 10 | 108.155 9 | 83.600 17 | 30.267 18 | 50.694 11 | 6.700 5 | 121.028 5 | <>052-05 | 5.75 4 | 5,00 3 | • | 7.50 3 | 5.00 3 | 4.25 3 | 3.25 2 | 1.75 1 | 8.75 4 | |
| | | | | 6.4000 34 | 8.5500 108 | | 9.6667 30 | | 4. 8 000 6 | | | 9.65217 | 0.8000 | | 0.16667 | 9.75990 | 91114 | 9.38462 | .42857 | 0.40000 | |
| | | 1 16.4444 | 1 22.0000 | 2 6.4 | 5 8.5 | 6 19.0000 | 9.6 | 3 10.2222 | 9 4 | 2 26.4444 | NH Fot-zg | 52 | 8 | 63 | 8 | 60 | 58 | 63 | 20 1 | 82 6 | |
| | -21-64 | 2.16667 | 2.50000 | 2.66667 | 3.00000 | 2.90909 | 4.25000 | 2.50000 | 2.50000 | 2.00000 | UOM-ZJ Houh | 21 2 | 7 18 | • | 8 18 | 6 14 | 4 14 | 4 10 | 4 14 | 10 19 | |
| | 0-1-F0 | 6 | • | <u>0</u> . | 4 | 6 | 2 | \$ | 6 | 10 | de-ofexy Fofexy | 23 17 | 20 13 | •] | 30 14 | 20 14 | 17 12 | 13 10 | 2 2 | 32 16 | |
| - | -Z2FEQ | 2.4286 | 2.0000 | • | 2.5000 | 3.7500 | 3.2857 | 14.0000 | 2.3333 | 2.4118 | | 15.4799 | 11.7220 | 7.1522 | 7.7969 | 13.1501 | 12.469a | 8.9775 | 7.7690 | 9.9848 | |
| - | 0-4-0 | 2.60000 1 | 2.71429 2 | 2.62509 3 | 1.66667 6 | 3.50000 0 | 2.40000 2 | 4.33333 0 | 3.00000 | 3.00000 5 | NOBE | + | • | 8 | | * * | | • | • | | |
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| • | -X1HZ- | 1.80000 | 3.00000 | 2.00000 | 5.00000 | 2.00000 | 6.00000 | 4.50000 | 5.00000 | 1.80000 | ¥2 < e o | | • | • | • | • | | • | | • | |
| 00 | 7874 7874 | ¥ 2 | ¥ 2 | 12 Y | N B | Υ 6 | 3 N | 8 У | N C | 6 Ү | ><* | 141.526 | 56.222 | 25.931 | 42.949 | 62.628 | 495.596 | 107.829 | 638.700 | 15.554 | |
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Table 3-5. Master Data Table (continued)

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| HOH-EGAA | ٠ | 16 | 18 | 31 | \$ | • | • | • | 38 | |
| ><24< | è | 109.908 | 16.333 | 6.679 | 13.240 | • | • | .• | 25.740 | |
| ender 4 | • | 26.500 | 12.333 | 9.330 | 929-2 | • | : : : • | • | 13. 140 | |
| ><≍0+ | • | 16.530 | 2.147 | 3.314 | 1.550 | • | • | • | 3.230 | |
| | | 4.500 | 2.833 | 2.770 | 2.220 | • | • | ÷ | 2.250 | |
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| Fog-Feme | 5.29832 5.29832 . | 17 12 | • | 19 21 | 12 14 | 20 | 22 | 5 | 16 14 | |
| Fog-Feme | 5 | 5.29832 17 12 | 8.16052 | 6.62007 19 21 | 6.62007 17 14 | 7.31322 20 | 8.16052 22 | 6.62007 | 5.99146 16 14 | |
| Fog-Feme | 5.29832 5. | 3.91202 5.29832 17 12 | 7.82405 8.16052 . | 6.62007 6.62007 19 21 | 6.62007 6.62007 17 14 | 7.31322 7.31322 20 | 7.82405 8.16052 22 | 6.62007 6.62007 15 | 5.99146 5.99146 16 14 | |
| Fog-Feme | 0.75000 5.29832 5. | 0.80000 3.91202 5.29832 17 12 | 7.82405 8.16052 . | 5 0.25000 6.62007 6.62007 19 21 | 15 0.75006 6.62007 6.62007 17 14 | 7.31322 7.31322 20 | 7.82405 8.16052 22 | 0.50000 6.62007 6.62007 | 14 0.70000 5.99146 5.99146 16 14 | |
| Fog-Feme | 15 0.75000 5.29832 5. | 16 0.80000 3.91202 5.29832 17 12 | 7.82405 8.16052 . | 0.25000 6.62007 6.62007 19 21 | 0.75006 6.62007 6.62007 17 14 | 7 0.35000 7.31322 7.31322 20 | 5 0.25000 7.82405 8.16052 22 | 10 0.50000 6.62007 6.62007 15 | 0.70000 5.99146 5.99146 16 14 | |
| Fog-Feme | 1.50 15 0.75000 5.29832 5. | 16 0.80000 3.91202 5.29832 17 12 | 7.82405 8.16052 . | 4.00 5 0.25000 6.62007 6.62007 19 21 | 1.50 15 0.75006 6.62007 6.62007 17 14 | 1.25 7 0.35000 7.31322 7.31322 20 | 0.75 5 0.25000 7.82405 B.16052 22 | 0.00 10 0.50000 6.62007 6.62007 5 | 4.75 14 0.70000 5.99146 5.99146 16 14 | |

Table 3-5. Master Data Table (continued)

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original page is of poor quality

| | ÷ | | | | | | | | | | | | | | | | | | |
|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-----|------------|--------------|--------|-----|---------------------------------------|----------|-------|------------|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-------------|-------|------------|--------------|-----------|----------|---------|------------|----------|
| 0 - | 5 | LO - | 5 | in. | 10 | 0 | 5 | 9 | 5 | HOUH4 | 2 | • | Ċ | - | 10 | , | 13 | | 0 |
| じれードムー | 2 | 4 | 3 | 4 | 3 | + | 3 | 4 | 0 | DEFECH | 8 | - | - | ið. | - | | 6 | e | - |
| 4F-14- | 3 | 3 | e | 3 | ŝ | 2 | 3 | 3 | 3 | | | | • | | - | | | 0 | N |
| UOZL- | 3 | 61 | 10 | 8 | 91 | 6 | 9 | 6 | t ~ | HOHEKX4 | U | - | M | 9 | 21 | LO. | 18 | 44 | 14 |
| | | - | - | | - | - | | - | | | 15 | 18 | 48 | - | 34 | 34 | 53 | 3 | 61 |
| OK-FA- | 10 | M | M | 2 | 3 | | 2 | М. | 5 | ガルズひです | | - | 4 | N | | 3 | | 3 | |
| アアリヨー | 0 | - | - | - | | 2 | | 2 | ŝ. | HOHHEdt | 5 | M | 15 | 28 | 32 | 4 | | 13 | 13 |
| じんてらじれる | | * | ŝ | - | 3 | 0 | 3 | 9 | 9 | | | | | | 7 - | | ÷. | * | 2 |
| NECOHOU | Ν. | 6 | 10 | - | c | 13 | 2 | - | 3 | 04 | -1.5 | 2 | 10 | 0 | b | CI | | Ţ | |
| UAFOUE- | | 5 | - | 3 | - | 4 | 5 | 4 | 2 | じまードA4 | 10 | ç | ¢ | 10 | 5 | 9 | 2 | 10 | 4 |
| • •. • | •• | | - | | • | 6 | | Ġ | 10 | ドアレムキ | 63 | 3 | ŝ | ŝ | Υ. | 3 | 3 | 2 | 2 |
| NOOKA | -9 | 9 2 | 69 | 40 | ũ | iii | 59 | Ň | iń | COXF4 | 9 | 9 | 8 | 9 | 9 | 9 | \sim | 8 | 9 |
| | - | 8 | - | 9 | ٢ | - | , mai | ٢ | - | | | | | | | _ | 5 | ø | 5 |
| | 5 | 9 | 9 | • | • | 0 | • | 0 | | OK-FB4 | 10 | LØ. | ୍ଷ | 9 | 4 | 4 | | | |
| 2. 4 1-6 | 6 | _ | 6 | • | | 6 | | | \$ | 新市山田市 | 3 | 2 | 3 | - | \$ | . | 0 | N. | <u>N</u> |
| 24-2 | 6 | 0 | | | 0 | | 6 | 3 | ē | X 臣 X B Y B Y B Y B Y B Y B Y B Y B Y B Y B Y B Y B S B S B S B S B S B S B S B S B S B S B S B S B S B S S S S S S S S S S S S S | 7 | Y | X | × | N | X | N | ~ | 7 |
| | | - | - | 6 | 6 | 6 | 6 | Ğ | 6 | Cod-zon | 4 | ~ | IQ. | 2 | ٢ | N | ۲ | 4 | 3 |
| 4 2 | 9 | 9 | Ģ | - | - | - | - | | - | FOUFW | F | - | 2 | - | ٢ | - | • | . 🕶 | - |
| 0 D d n | . 🛏 | 9 | النتعج | • | • | - | I | - | - | DZFECXO | Q | -44 | Q | LO . | 2 | ¢1 | 9 | 3 | 3 |
| OD#+ | | | 9 | 9 | 0 | - | ٢ | - | 9 | HOHEKAN | æ | 10 | • | 8 | 6 | ŝ | 5 | ŝ | 4 |
| OD M O | ٩ | | 9 | ٩ | | 9 | 9 | - | THE | | | | Ē | | | | | | |
| じりれる | 9 | \$ | 0 | - | 9 | 9 | 9 | ٩ | S | MAXAHO | 19 | 23 | 19 | 121 | 52 | Ν | 36 | 17 | 63 |
| - Som - | • | \$ | 9 | ٩ | 0 | 0 | 9 | ٢ | 9 | | | 6.0 | | - | 44 | | | | • |
| | 6 | 6 | 0 | 6 | Q | Q | ~ | ю | • | | 53 | 14 | 24 | 22 | 6 | 10 | 36 | 6 | 10 |
| | • | | ÷, | - | | | | | | HOHHZOM | 0 | | 0 | 54 | | | - | | |
| ACB. | 2 | 0 | 2 | 2 | 3 | 3 | 4 | 4 | . 61 | 0.00 | ın, | 10 | ß | 10 | ¢ | 10 | ٢ | 10 | l0 |
| HZSH | 4 | - | - | - | 9 | i i i i i i i i i i i i i i i i i i i | ŝ | | şiini | じューアムの | 4 | 4 | 4 | in. | 3 | 5 | 2 | 9 | 2 |
| | ý. | - | | - | 9 | - | 9 | T | 4 | F 140 | 3 | ŝ | 0 | 3 | ¢ | e | ŝ | 3 | 2 |
| 2-0 | - | = | | - | - | - | - | - | - | UOZEO | 10 | 9 | 6 | • | 3 | 10 | 9 | ٢ | • |
| | - | - | N | - | 4 | 4 | 10 | 3 | in | | |) and | | - | | | | | |
| | | 8 | ~ | - | 2 | | ср ср | 1 | 2 | じれード曲の | 9 | 4 | 9 | ŝ | 4 | Ŷ | r• | 9 | 0 |
| 0 O | 2 | | | | | | | | | あてしほう | 2 | . လ | 2 | .0 | 0 | 3 | - | 3 | 0 |
| | 20 | 50 | 50 | 200 | 50 | 200 | 50 | 500 | 50 | | ¥ | N | ¥ | N. | 7 | · N | 7 | N | N |
| | | | | 0 | | Q1 | | 12 | | DOGHZON | - | ٢ | 'yaani | • | 0 | 9 | 0 | - | ٢ |
| S H H S | | ۲ | • | da l | • | | 6 | • | 6 | HOUFN | - | • | _ | | 0 | • | 3 | | • |
| V2 | 20 | 750 | 200 | 50 | 750 | 500 | 400 | 2500 | 1500 | DZHEKKN | | ő | e | 4 | 4 | | 10 | 4 | e |
| | | Le. | 64 | | .[- | - | ¥. | ត័ | 1 | | ,- - | | | | ف | 4 | 4 | 4 | 4 |
| | 20 | • | ٢ | 9 | 9 | Q | 0 | G | 9 | HOHEAXA | ŝ | ġ | 4 | • • | • | ч. | - | A . | м. |
| 이 나는 문문에서 | 5 | 200 | 200 | 250 | 750 | 500 | 400 | 7500 | 500 | M J D X P X | 48 | 54 | 33 | 30 | 55 | 94 | 69 | 18 | N. |
| | | ÷. | ••• | | - | <u> </u> | - | 2 | - | | 4 | 0 | 2 | 3 | ର୍ଷ | • | 10 | 1 | t• |
| | - | | | N | - | 2 | - | ю | - | | | | 6 | 5 | | | 15 | ~ | 9 |
| | | | | | | | | 1.1 | | FOFTZON | 25 | 10 | | 16 | 19 | 13 | 23 | • | - |
| S HE | Ţ | 4 | 2 | | 4 | 13 | | 9 | 10 | 50 | LD. | G | 2 | • | • | 9 | LO | 0 | G |
| | - | io. | 0 | 4 | 4 | 6 | 3 | .0 | 5 | UE-FAN | 5 | 4 | - | 9 | 9 | 9 | 3 | 4 | 3 |
| HAF-0 | 9 | • | • | • | ٥ | • | • | • | 6 | 5 F L H | | | | 3 | 3 | 0 | 3 | e | 2 |
| | | 6 | 6 | 6 | 6 | 0 | 6 | _ | 6 | CORFU | 0 | - | N | ف | ف | 5 | 10 | 8 | lo. |
| HAF0 | | _ | - | | | | | _ | - | | •• | | 4 - | - | . – | | | | |
| H < | - | 1 | 9 | 8 | • | - | 9 | | - | 02-1-80 | ¢ | 3 | 10 | ø | | Ŷ | 4 | 4 | 3 |
| RAHN | | T | - | 9 | - | - | - | | - | ド て し り こ | 3 | 3 | ŝ | - | | 3 | 3 | 3 | 0 |
| RAF 0 | G | T | 0 | 9 | | 0 | . © | and a | , | Max- | 2 | · >- | Я | 7 | × | 3 | þ. | N | 3 |
| R <fid< td=""><td>•</td><td>-</td><td>0</td><td>- vend</td><td>-</td><td></td><td>G</td><td>-</td><td>1</td><td>002-20-</td><td>6</td><td>-</td><td>Ň</td><td>, IQ</td><td>50</td><td>-</td><td></td><td></td><td>0</td></fid<> | • | - | 0 | - vend | - | | G | - | 1 | 002-20- | 6 | - | Ň | , IQ | 50 | - | | | 0 |
| KAH4 | • | 9 | 0 | 0 | • | ٢ | 0 | 0 | 9 | -404 | | - | | | 2 | - | ୍ଷ | 0 | |
| E <fo< td=""><td>9</td><td>9</td><td>9</td><td>0</td><td>0</td><td>9</td><td>٢</td><td>0</td><td>6</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td></fo<> | 9 | 9 | 9 | 0 | 0 | 9 | ٢ | 0 | 6 | | | | | | | | | | |
| RAFA | | | • | • | ٩ | 6 | ö | G | 0 | DZHZKX- | | | | <u></u> | 6 | | 4 | 2 | 4 |
| E <f-< td=""><td></td><td></td><td>0</td><td>6</td><td></td><td>•</td><td></td><td></td><td>6</td><td>FOFEKA</td><td>iņ.</td><td>-</td><td>l0</td><td>10</td><td>4</td><td>3</td><td>10</td><td>C1</td><td>4</td></f-<> | | | 0 | 6 | | • | | | 6 | FOFEKA | iņ. | - | l0 | 10 | 4 | 3 | 10 | C1 | 4 |
| CHEF. | 0 | | - 20 | | 8 | | | | | ZAXAH- | 63 | 10 | 53 | . L <u>D</u> | 64 | 61 | 31 | 5 | |
| | | | | | | | | | | | | 4 | | 6 | | N | .0 | 0 | |
| NFDAWE | 15 | 52 | 53 | 10 | 55 | 56 | 57 | 58 | 56 | FOF-20- | - 9 | | 01 | | Ψ. | | H | - T | 4 |
| | - | 22 | 23 | 5 | 22 | 26 | 22 | 28 | 26 | ······································ | - | 52 | 23 | 4 | 19 | 26 | 25 | 28 | 60 |
| NFDORF | | , ini | - L D | 5 | 5 | 10 | 10 | 10 | . 13 | NFDOWNH | 10 | 10 | . 10 | 5 | 13 | 5 | 17 | | 5 |
| | | | | | | | | | | | | | | | | | | | |

Table 3-5. Master Data Table (continued)

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| | | | | | | | | | | OF | RIGINAL | PAG | e 19 | 5 | | | | | | | | |
|-------------|--------------|---------|---------|---------|---------|---------|---------|---------|-----------|------------|-------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-------------|------------|-----------|-----------------|-------------------|------------|------------|-----------|------------|------------|---|
| ¥84< | 24 | 4.4706 | 5,000 | 8.6000 | 4.2857 | 5.3433 | 7.9206 | 5.9250 | 10.7059 | 80F | POOR | QUAI | | 10 × 10 | 2.23077 | 2.67857 | 2.30000 | 2.93333 | 2.19231 | 3.41667 | 2.26316 | |
| 24 | E 0 | 42.613 | 45.451 | 36.522 | 16.123 | 421.250 | 2.456 | 55.349 | 24.500 | 394.933 | Av | -074 | 0.217391 | 9.399998 | 0.163462 | 0.089286 | 0.083333 | 0.160000 | 0.046154 | 0.150000 | 0.126316 | |
| Z N4 | ZO | 6.3913 | 6.7143 | 6.5000 | 5.8636 | 17.0000 | 4.3000 | 6.2778 | 9.6667 | 15.6000 | <>0H0 |) z | 18.75 | 8.75 | 14.50 | 18.75 | 17.25 | 8.80 | 22.80 | 8.20 | 8.60 | • |
| A 4 | ≝ २३ | 127.33 | 63.61 | 50.94 | 59.36 | 26.58 | 626.58 | 146.92 | 1638.00 | 696.67 | <>000 | | 9 4.50 | 0 2.75 | 5 5.00 | 5 4.00 | 5 4.50 | 0 2.20 | 0 1.60 | 09-1-60 | . 1.60 | |
| K N4 | ZN | 7.4909 | 8.5000 | 7.7778 | 6.8125 | 5.0421 | 18.0769 | 9.5660 | 28.6000 1 | - 28. 3333 | <>0F | -005 | 3.75 1.50 | 25 0.50 | 4.00 2.25 | 4.25 1.25 | 25 2.25 | 40 0.60 | 4.00 2.00 | 2.40 0.60 | 2.80 0.80 | |
| | | | | | | | | | | | | - OF | 5.75 3. | 2.50 2.25 | 6.50 4. | | 7.50 5.25 | 3.60 2.40 | 10.40 4. | 2.40 2. | 3.80 2. | |
| >4 | H | 396.456 | 35.583 | 63.378 | 84.944 | 493.778 | 26.810 | 90.333 | 34.286 | 14.835 | | | | | | | | | | | | |
| Z N A | >* | 15.3000 | 6.2500 | 10.6000 | 7.778 | 13.4444 | 12.1429 | 9999-6 | 7.5000 | 5.2857 | LOL | NF | 6.86957 | 1.2000 | 9.65385 | 9.35714 | 0.33333 | 999999 | 6.23077 | 0.75900 | 3 0.63158 | |
| | | 2 | 6 | e | 4 | 13 | - | 13 | 0 | 8 | | 1-29 | 75 | 35 | 58 | 22 | 69 | 4 | 114 | 41 | 43 | |
| | | 6 | 8 | 9 | | | 6 | | N | * | UOE | (-zd | 18 | 11 | 20 | 16 | 18 | 11 | 8 | 120 | 8 | |
| | # + | 3.4000 | 7.00000 | 2.14286 | 3.11111 | 1.64235 | 2.80000 | 2.2222 | 5.66667 | 1.85714 | 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - | -001 | 15 6 | 9 | 16 9 | 2 | 13 9 | 12 3 | 20 10 | 12 3 | 14 4 | |
| 0 | r ⇔ | 0 | | 4 | 3 | I | - | 6 | 6 | T | DEHOL | | | | | 12 | | | | | | |
| | # 0 | 2.87500 | 2.80000 | 2.40000 | 2.75000 | 3.00000 | 3.33333 | 2.40000 | 3.00000 | 2.50000 | For Ram- | H A H | 8.3905 .23 | 6.6161 10 | 8.3694 26 | 6.1849 28 | 10.4076 30 | 10.6121 15 | 7.6907 52 | 13.9681 12 | 14.1125 19 | |
| D -F | 10 | 2 | 9 | - | 0 | 0 | - | 6 | 9 | | U. | ONE | ÷ | 4 | ÷. | | | 10 | | 10 | 10 | |
| | ∝ 0 | 5.00009 | 3.33333 | 2.25000 | 2.66667 | 3.16667 | 3.25000 | 1.78571 | 1.75000 | 1.50000 | | < # 19 | • | • | • | • | ٠ | 129.15 | 158.02 | 249.13 | 142.96 | |
| 0-4 | — — | N | 9 | 0 | Ň | T | • | - | 8 | • | | | | | | | | 33 | 60 | 000 | 42. | |
| | • • • • • | 2.00000 | 4.0000 | 2.00000 | 1.80000 | 2.25000 | 2.33333 | 2.60000 | 4.00000 | 3.50000 | X | | * | • | • | • | • | 12.8235 | 8.0909 | 13.0000 | 7.6774 | • |
| HEAH | | ¥ | • | | | | | | | | | المد في بقر | 15.640 | 35.333 | 400 | 19.693 | 82.233 | 75.456 | 28.738 | 136.471 | 28.859 | |
| CO4-X | 5. T | 2 | N 6 | 7 | ×. | | 3 | 6 N | 3 N | 2 | 2 | | 15. | 35. | 164.400 | 61 | 82. | 75. | 28. | 136. | 28. | |
| 0 HDAM | | 12 | 25 | 53 | 12 | 10 | 26 | 22 | 28 | 65 | 0 H D B | SEL. | 31 | 52 | 23 | 1 | 22 | 92 | 57 | 38 | 23 | |
| 1 H I I I | | | | | | | | | | | | | | 1.1 | 11 | . ¹ 1. | | | | | | |

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Table 3-5. Master Data Table (continued)

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| N M L M O | e | 1 | 3 | 3 | 8 | - | | - | - | |
|---------------------|----------|----------|----------|----------|----------|----------|------------|----------|----------|--|
| YN | - | ymt, | - | ٩ | , | - | = | T | - | |
| >-x0<-2 | 12.7 | 0.86 | 2.57 | 3.14 | 6.30 | 1.00 | 2.25 | 0.00 | 3.25 | |
| M-NG4-H | 4.50 | 1.89 | 2.71 | 2.00 | 4.50 | 4.50 | 5.25 | 4.00 | 4.00 | |
| DZ-OC-EL | Ŷ | 9 | Ņ | 2 | 10 | 4 | : 4 | 4 | 8 | |
| HOH4-EL | 8 | ¢ | Ν | 8 | 9 | 4 | 4 | 4 | 9 | |
| HOH-EGAA | 36 | 17 | 19 | 16 | 22 | 18 | 21 | 16 | 36 | |
| ><24< | 12.910 | 14.500 | 223.240 | 38.550 | 5.340 | 114.250 | 39.000 | 44.250 | 179.759 | |
| Xuerte | 10.625 | 8.660 | 20.700 | 9.625 | 17.167 | 17.250 | 20.500 | 19.750 | 15.670 | |
| VAKUF | 1.310 | 1.370 | 13.990 | 3.857 | 1.520 | 4.235 | 10.030 | 1.862 | 076.1 | |
| Magaah | 2.780 | 2.390 | 3.890 | 3.250 | 2.150 | 3.900 | 4.333 | 2.560 | 2.330 | |
| FORTHAR | 16 | 15 | 14 | 20 | 17 | 21 | 10 | 19 | 11 | |
| FUM-FARE | 61 | 19 | 20 | 19 | 18 | 20 | 24 | 23 | 15 | |
| 66700 | 3.91202 | 7.31322 | 5.29832 | 6.62007 | 6.62007 | 7.31322 | 5.99146 | 8.92266 | 7.31322 | |
| 88F-100 | 3.91202 | 6.62007 | 5.29832 | 3.91202 | 6.62007 | 7.31322 | 5.99146 | 7.82405 | 7.31322 | |
| RFOHUOK | 1.00 | 0.60 | 0.85 | 0.50 | 0.50 | 0.48 | 0.48 | 0.36 | 0.48 | |
| Force | 20 | 12 | 21 | 10 | 10 | 12 | 12 | 6 | 12 | |
| <>00-4FF | 2.00 | 0.25 | 2.50 | 2.75 | 4.25 | 0.60 | 5.40 | 0.00 | 1.00 | |
| □ - 1, i- i- | | - | 10 | 11 | 21 | 3 | 32 | 0 | 10 | |
| <>0-24F4 | 0.815217 | 0.875000 | 0.557692 | 0.669643 | 0.575000 | 0.586667 | 0.438462 | 0.683333 | 0.452632 | |
| SFDCHSF | 21 | 22 | 53 | 45 | 133 | 36 | 22 | 28 | 26 | |

Table 3-5. Master Data Table (continued)

| . | 5 | • | | -01 | 6 | • | N | 10 | ю | FOUF4 | - | - | - | - | 2 | - | ÷ | | Ð |
|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|------|------------|-----|-----|------------|----------------|-----|------|--------------|----------------------------------------------------------------------------------------------------------------------------------|-------|------------|-----------|------------|-----|------------|----------|----------|-----------|
| | - | 0 | | N | ~ | D | 5 | N | 0 | DZHEKX | * | 3 | 0 | N | 6 | ~ | e | - | |
| ->LHH | ö | | | | - | 0 | | 8 | | FOFE <x4< td=""><td>न्तुः</td><td>3</td><td></td><td>N</td><td>6</td><td>N</td><td></td><td>N</td><td>4</td></x4<> | न्तुः | 3 | | N | 6 | N | | N | 4 |
| UOZL- | 6 | 5 | • | Ň | 10 | 6 | | | | | | •• | | | • | | •• | • ** | • |
| | | • | • | | | | | | 10 | X <x0f4< td=""><td>32</td><td>50</td><td>30</td><td>12</td><td>28</td><td>18</td><td>8</td><td>29</td><td>\$</td></x0f4<> | 32 | 50 | 30 | 12 | 28 | 18 | 8 | 29 | \$ |
| UH-HA- | 2 | 3 | • | N | - | 9 | È. | 3 | 2 | | 63 | 64 | 63 | | କ | | | 51 | • |
| 14 F 12 B - | 2 | 3 | • | 63 | 3 | 0 | - | 2 | - | | - | 6 | ► | 2 | 28 | • | 4 | 16 | 10 |
| 0 4FOURS | | ю | 3 | - | N | N | N | | Ċ | HOHHZO4 | - | | | | ରା | | | | |
| DAHODEN | | - | 4 | ë | 2 | | 2 | | | 04 | 2 | • | 63 | 0 | 5 | 10 | 0 | 0 | 0 |
| DAFNOE- | 3 | | j. | - | 5 | 2 | 4 | 5 | ÿ | 04-14+ | 1 | 0 | C1 | 3 | 0 | - interior | 10 | ŝ | 4 |
| | | | - | • | • | | • | | - | ドモレム4 | ŝ | ŝ | 3 | 3 | 3 | 3 | 3 | ŝ | 3 |
| SCORE | 25 | 65 | 65 | 53 | 45 | 35 | 40 | 50 | 60 | | ٢ | 9 | | 9 | 2 | O | | 8 | 9 |
| あれてら | - | , i | - | - | -i | - | - | - | , | CONT4 | - | | • | ., | | _ | | | |
| | ۵ | 9 | 3 | 9 | 0 | 6 | 0 | 9 | \$ | 04-F34 | - | | 19 | | - | | N | 3 | 13 |
| 8. 6 - 6 | | 6 | 6 | 9 | • | | 0 | • | 9 | | .0 | .00 | 9 | | 3 | | 0 | 2 | |
| 24-01 | 6 | 6 | | 6 | 6 | ŝ | 6 | 6 | | X H X S | 7 | N | Z | ž | N | ž | | N | z |
| | ō | 6 | 6 | ō | 6 | 6 | 6 | 6 | ő | COHHZON | 4 | Ņ. | 9 | ŝ | T | 3 | 9 | T | **** |
| UDE0 | | - | - | | - | - | - | - | - | HOUFS | | 2 | 9 | - | ÷ | - | ٩ | - | - |
| | | | | | - | 14 | - | | - | DZFECIO | 5 | 9 | 3 | | -11 | ŝ | 4 | Ν | 9 |
| | 0 | ٩ | | 0 | 9 | 9 | ٢ | S | ٩ | HOHEAND | 3 | lo, | ĊV. | 4 | • | 4 | 10 | 18 | 10 |
| じりまの | 0 | 9 | ° 🗢 | 0 | 9 | 4 | 0 | 9 | \$ | | | | | | | | | | |
| じりほう | 9 | 9 | ¢ | 9 | 9 | ¢. | 9 | 9 | 9 | MAXUFO | 2 | 2 | 36 | 21 | 33 | 91 | 44 | 30 | 13 |
| UDE- | . 🗢 | 9 | ٢ | 9 | 9 | ۵. | 0 | 9 | 0 | | | - | | | | | | | |
| HINT | 8 | Ġ. | 2 | 6 | 9 | 16 | 6 | 9 | 9 | HOFHERD | 11 | 31 | LO. | 10 | 10 | 8 | 25 | 31 | 16 |
| | | | | | _ | | | | _ | | | | 6 | 10 | 2 | | | 0 | 6 |
| A Q M | 0 | 0 | 5 | - | 0 | 0 | - | 2 | 0 | 00 | | | - | | | 0 | - | | - |
| HUZH | 2 | Ņ | 0 | 2 | 3 | 5 | - | 4 | 9 | UM-HAQ | C1 | 5 | 13 | Ъ, | 10 | | N | i0 | 4 |
| | 2 | 9 | i) | 2 | 4 | 5 | 9 | 9 | 9 | 2714F | 3 | 3 | 3 | ŝ | 2 | 63 | 2 | ŝ | c, |
| 8 V | 2 | | T | 2 | ŝ | , 1 | | - | - | COZIC | 9 | 8 | 8 | 10 | 4 | 8 | 5 | 2 | N |
| m H | 0 | 3 | 4 | 2 | 3 | io. | - | yest | 2 | U E H H B B | N | Ν | 4 | 10 | Ś | N | N | 4 | 0 |
| S S S S S S S S S S S S S S S S S S S | N. | 0 | 0 | 3 | 2 | 01 | 2 | C) | 2 | 01-F180 | 7.5 | - | - | 8 | 2 | | - | 2 | 6 |
| mere | | 20 | 50 | 50 | ٩ | 9 | • | 20 | 20 | | | | | | | | | | |
| | 10 | 5 | ŝ | C) | 200 | 5 | 6.0 | 10 | 5 | X 単 X A | N | : Z | Z | * | N | Z | N | N | Z |
| | | | | | | | | | | CORHZON | ۹, | 2 | 2 | 2 | 0 | 5 | a | 4 | 9 |
| S H F S | 200 | 260 | 200 | 200 | 260 | 20 | 200 | 50 | 50 | HOUHN | 9 | ŝ | | 0 | ÷ | - | 8 | 4 | 0 |
| | Ň | Ñ | Ň | Ň | Ň | | ลั | -, | | DZFE4XO | 3 | 4 | 2 | 5 | - | | 4 | 4 | ¢, |
| -H- | | - | _ | ~ | • | • | - | _ | _ | FOFEAN | 13 | 8 | 2 | 8 | - | - | 50 | 12 | 4 |
| | 200 | 200 | 200 | 200 | 200 | 59 | 200 | 20 | 50 | | | - | | | | | | | |
| | . 64 | e i | 6N) | n. | ୍ୟ | | 5 | | | MAX0F0 | 40 | 44 | 19 | 75 | • | 4 | 49 | 33 | 99 |
| en e | | | | | | | | | | | _ | | _ | - | | | | | |
| | - | T | - | ÷4 | 2 | 1 | | | | HOHHZON | 16 | 15 | c, | . 0 | 2 | μ¢. | 35 | 20 | 10 |
| SHE | 2 | 2 | 2 | 2 | 2 | - | 2 | - | | 00 | 6 | | 0 | 10 | • | 0 | 0 | 0 | 6 |
| | 2 | 0 | 2 | 2 | 2 | Ţ | 2 | | Ţ | ちょうぼい | N | 4 | 10 | 3 | 3 | 3 | Q | 4 | 10 |
| HAH-0 | c | 0 | 0 | | 6 | | 6 | 5 | | 5 FLAN | | | Ö | | | | | с с | .0 |
| | | | | | | | | - | | UOZLO | | ÿ | | 10 | 5 | 6 | | | - |
| EAF 0 | | | 0 | | | • | | | | UURHN | 6.2 | • | - | | 47 | er. | 10 | N | - |
| 847-8 | | | 9 | 0 | 9 | | - | 0 | | OH-FEN | 3 | ŝ | 5 | છ | . = | 9 | - | - | 2 |
| RAFr | | H | T | | . – | T | | T | - | F FJB8 | e | 3 | 0 | 3 | | 0 | 3 | 6 | - |
| 44Fo | | | 0 | 9 | 6 | 0 | 0 | 9 | 0 | X8×- | 7 | × | | 7 | 7 | N | | × | 7 |
| RAF D | | | 9 | 0 | 0 | 9 | 9 | 0 | ø | UOZ-ZO- | | * | • | 10 | j. | | 5 | 2 | 4 |
| K <f4< td=""><td>G</td><td>4</td><td>0</td><td>0</td><td>•</td><td>٢</td><td>0</td><td>۰</td><td>9</td><td>6006-</td><td></td><td>,</td><td></td><td><u> </u></td><td>N</td><td>_</td><td>ล</td><td>2</td><td>-</td></f4<> | G | 4 | 0 | 0 | • | ٢ | 0 | ۰ | 9 | 6006- | | , | | <u> </u> | N | _ | ล | 2 | - |
| R <f c<="" td=""><td>9</td><td>•</td><td>9</td><td>9</td><td>0</td><td>٩</td><td>•</td><td>9</td><td>0</td><td>DZHECK-</td><td></td><td>4</td><td></td><td>N</td><td></td><td>_</td><td></td><td></td><td>_</td></f> | 9 | • | 9 | 9 | 0 | ٩ | • | 9 | 0 | DZHECK- | | 4 | | N | | _ | | | _ |
| # <+0 | • | 0 | 0 | 0 | ٢ | 9 | 0 | ٢ | 0 | | | | • | | | | | | |
| E < | 0 | 0 | 0 | 0 | • | ¢ | 0 | • | 6 | HOHEAN | | N | | 0 | 6 | | 2 | 10 | |
| CHEF | | | Ň | 2 | N | 2 | Ň | | 2 | EXX01- | . 6 | 42 | 0 | 57 | 22 | 4 | 10 | 25 | 9 |
| | | | | | | | | | | HOH-Zd- | 0 | 2 | 0 | N | 50 | 9 | 9 | 10 | * |
| NEDORE | 99 | 61 | 62 | 63 | 64 | 65 | 66 | 29 | - 6 B | | | | | | | | | | |
| DEDOWER | | | N | 19 | | 12 | 9 | 2 | | OFDANEF | 9 | | S | 3 | | 10 | 9 | 2 | |
| | | | | | . 0 | 9 | 0 | | V | | | | • • 🗳 • | | | | : 🖉 - | | |

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| | 1.2727 | 6.7778 | 8.8571 | 9.0000 | 5.6071 | 5.6667 | 6.000 | 5.4500 | 12.7009 | | 3.15305 | 2.03030 | | .75000 | 2.53571 | 3.12500 | 2.00000 | 1.83333 | 2.10526 |
|-----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|----------|---------|------------|------------|----------|---------|---------|---------|---------|-------------------------------------------------------------------------------------------------------------------------------------------------------------------|------------------|------------|---------|-----------|------------|----------|-----------|------------|---------------|
| Ku <rt< th=""><th></th><th>Ŵ</th><th>æ</th><th>ġ</th><th>ЪĢ.</th><th>ġ</th><th>ø</th><th>10</th><th>12</th><th>-zefet</th><th>3.11</th><th>2.0</th><th>•</th><th>1.7</th><th>2.5</th><th>3.15</th><th>2.0</th><th>1.8</th><th>2.1(</th></rt<> | | Ŵ | æ | ġ | ЪĢ. | ġ | ø | 10 | 12 | -zefet | 3.11 | 2.0 | • | 1.7 | 2.5 | 3.15 | 2.0 | 1.8 | 2.1(|
| ><40 | 28.473 | 34.170 | 195.700 | 55.511 | 80.124 | 25.268 | 117.417 | 43.531 | 10.696 | | 9. 184615 | | | 0.175000 | 9.985714 | 0.250000 | 0.022857 | 0.042857 | 0.073684 |
| ZNAZ0 | 8.4545 | 6.6452 | 14.8000 | 9.2000 | 9.5333 | 6.1250 | 9.4000 | 7.7419 | 5.1875 | 47049 2-404042 | 8.2000 0 | 13, 4000 | 3.7500 | 5.6000 0. | 14.2000 0. | 5.0000 0 | 14.0000 0 | 15.4000 0. | 8.0000 |
| ><=0 | 132.196 | 130.210 | 72.333 | 416.500 | 8.000 | 34.200 | 85.563 | 125.039 | 390.678 | 470002-z | 2.00 | 2.20 1 | • | 2.20 | 2.80 | 3.00 | 3.00 | 1.60 | 1.60 |
| . K | 14.0625 | 10.9333 | 9.3333 | 25.3333 | 7.0000 | 8.2000 | 6.2857 | 10.2590 | 16.3000 | 4>9F00F | 20 1.0 | 1.4 | • | 60 1.0 | 90 1.2 | 10 0.B | 60 2.2 | 50 1.6 | 50 0 4 |
| Z ə <z<i>n</z<i> | | | <u>u</u> , | | | | | | | 4900X-0H | 2.20 | 3.40 | • | 2.60 | 3.00 | 1.40 | 2.69 | 3.60 | 2.60 |
| 74X-1 | 4.333 | 183.545 | • | 361.619 | 42.806 | 1.000 | 16.390 | 55.656 | 12.000 | <>UF0FdX | B 2.6 | 979 | • | 9.2 | 2 5.6 | 0 1.6 | 9.2.6 | 9 8.4 | 2 3.8 |
| e sper | 6.6667 | 10.5000 | • | 19.5714 \$ | 7.3846 | 3.0000 | 6.5000 | 6.1000 | 4.0000 | NF | 0.92308 | • | • | 0.87500 | 0.42857 | 1.25999 | 0.11429 | 0.21429 | 0.36842 |
| 0-1 | 6 | 9 | 9 | 0 | 4 | 6 | , G | | - | FOF-ZO | 41 | 29 | 12 | 28 | 12 | 25 | 92 | 22 | 40 |
| | - | - | ŝ | 9 | | 9 | 2 | | 9 | UOE-20 | 10 | 11 | • . | IT | 14 | 15 | 15 | 8 | l) |
| | 2.75000 | 3.00000 | 2.33333 | 1.00000 | 3.11111 | 4.50000 | 1.33333 | 2.28571 | 2.50000 | HOOH | 1 | 2 21 | • | 13 5 | 15 6 | 4 | 13 11 | 18 8 | 13 2 |
| 0-4-6 | 3 | • | \$ | 0 | ŋ | - | 9 | II | 4 | PE-OFEEX | 13 . 1 | | • | 16 1 | 28 1 | | 32 1 | 42 | 1 61 |
| TELFER | 3.65667 | 2.06667 | 2.50000 | 2.50000 | 1.66667 | 2.00000 | 2.50000 | 1.72222 | 1.60000 | KPR44 | 10.1141 1 | 8.7141 33 | 10.9968 | 15.7762 1 | 7.3813 2 | 5.7479 | 7.0464 3 | 7.3855 4 | 9.5469 1 |
| 2-1-R | 2 | 4 | \$ | 3 | 6 | • | 16 | 8 | - | NORS | 10 | 10 | 4 | 5 | ß | 1D | 5 | IJ | ŝ |
| - Z L H K N | 3.20000 | 1.87500 | 1.50000 | 1.12500 | 2.99990 | 5.00000 | 1.75000 | 1.66667 | 2.50000 | ><#0 | 137.90 | 223.58 | 355.10 | 192.21 | 52.78 | 35.51 | 268.76 | 5215.06 | 159.90 |
| 0-4 | • | 8 | • | • | 4 | • | 6 | 61 | 9 | | 333 | 138 | 357 | 333 | 996 | 692 | 346 | 939 | 414 |
| | 3.00000 | 1.71429 | - | 3.50000 | 2.88889 | 3.00000 | 3.00000 | 2.00000 | 4.00000 | xu <ro< th=""><th>14.5833</th><th>9.1538</th><th>15.7857</th><th>15.7333</th><th>6.9900</th><th>7.0769</th><th>11.3846</th><th>34.3939</th><th>10.2414</th></ro<> | 14.5833 | 9.1538 | 15.7857 | 15.7333 | 6.9900 | 7.0769 | 11.3846 | 34.3939 | 10.2414 |
| -zihe- | | | • | ю. | N | | | | | | 618 | 42.694 | 143 | 18.000 | 321 | 24.250 | 4.667 | 28.366 | 011 |
| AMX-200 | 20 20 | ж е | 2 | I. | 2 | * | Z. | N | N 6 | ><=+ | 116.618 | 42. | 103.143 | 18. | 32.321 | 24. | 4 | 28. | 242.011 |
| | Ø | 5 | Ŋ | 23 | | 10 | | 2 | 89 | MLANDL | | | Ŋ | 5 | | ß | 9 | 2 | B |

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Master Data Table (continued)

Table 3-5.

| 0 H L H C | 3 | - | n | 12 | 5 | | | . | S | |
|--------------------------------------------------------------------------------------------------------------------------------------------------------------------------|----------|----------------|-----------|----------|------------|----------|----------------------|----------|----------|---|
| | - | - | - | - | ÷ | 1 | , - , | T | T | |
| >-xg<-z | 1.870 | 2.900 | 4.300 | 9-999 | 069.0 | 1.570 | 9.640 | 466.1 | 4.926 | |
| X-xo | 2.670 | 3.280 | 3.600 | 6.000 | 4-788 | 3.280 | 3.750 | 5.330 | 3.250 | |
| DX-OK-EL | Ņ | M | 10 | 3 | đ. | N | ø | e | 0 | |
| FOR4-EL | 9 | N | LØ. | 3 | œ١. | Ν | 8 | Ċ. | * | |
| HOH-ZOAA | 16 | 23 | 18 | 18 | 4 3 | 23 | 30 | 16 | 13 | |
| ><=L< | 44.650 | 109.900 | 87.299 | 10.330 | 43.998 | 33.619 | 39.696 | 5,333 | 243.667 | |
| Ku <ka <<="" th=""><th>12.500</th><th>12.719</th><th>17.200</th><th>18.670</th><th>13.570</th><th>12.570</th><th>13.625</th><th>13.330</th><th>19.500</th><th></th></ka> | 12.500 | 12.719 | 17.200 | 18.670 | 13.570 | 12.570 | 13.625 | 13.330 | 19.500 | |
| | 2.267 | 8.930 | 5 .670 | 14.490 | 1.597 | 3.976 | 3.685 | 25.850 | 8.141 | |
| ZUCKOF | 4.0000 | 3.7490 | 3.4400 | 5.6110 | 2.7006 | 3.3900 | 2.6670 | 5.8759 | 4.8460 | |
| HOR-HAR | • | * | ٠ | 13 | 21 | 10 | 24 | 13 | 15 | |
| FOR-FERE | 33 | 14 | ٠ | 18 | 6 | 17 | 22 | 11 | 16 | |
| 66744 | 5.29832 | 5.29832 | 5.29832 | 5.29832 | 5.29832 | 3.91202 | 5.29832 | 3.91202 | 3.91202 | |
| 004400 | 5.29832 | 5.29832 | 5.29832 | 5.29832 | 5.29832 | 3.91202 | 5.29832 | 3.91202 | 3.91202 | |
| r Fofuor | 0.480000 | .• | 0.100000 | 0.569909 | 0.480090 | 0.40000 | 0.160000 | 0.360000 | 0.280696 | |
| Foruor | 2 | ÷ | 0 | 14 | 12 | 10 | Ŧ | 6 | 2 | |
| <>00-4FF | 0.40 | 3.20 | • | 0.69 14 | 2.60 | 0.20 | 4.40 | 4.63 | 1.20 | |
| | N | 16 | • | 69 | 13 | Ŧ | 22 | 24 | 9 | |
| <>U-ZAFE | 0.630769 | 0.406061 | - <u></u> | 0.350000 | 0.507143 | 0.625900 | 0.400000 | 9.366667 | 0.421053 | |
| OFDONEF | | 19 | 62 | 63 | 49 9 | 65 | 99 | 29 | 68 | 1 |

Table 3-5. Master Data Table (continued)

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| | | | | | | | | | | OF POOR QU | ALI | ļ T | | | | | | | | |
|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|----------|------------|-----|------------|----------|-------------|--------------|------------|----------|----------------------------------------------------------------------------------------------------------------------------|----------------|---------|----------|------------|------------|------------|---------|----------------|------------------|---|
| UOZLN | ¢ | N . | M | 8 | 10 | n | 3 | Ģ | M | ••• | - | - | 2 | N | 2 | 2 | | | 2 | |
| OH | . | - 100 | 3 | 5 | 0 | 10 | + | 10 | 9 | | 3.00000 | 2.0000 | 1-50000 | .66667 | 2.0000 | 8.0000 | | | 2.8009 | |
| r F J B O | 3 | Ċ1 | 3 | 3 | 3 | 3 | 2 | 0 | 3 | | ð | ಿಕ್ಕೆ | ធុ | ভ | ē | ā | | | ā, | |
| XH>- | Z | X | | | | | | | | | 3 | 0 | | - | 2 | | | | 0 | |
| Uod-zd- | 3 | 9 | ٠ | . . | | | | .* | | 8-4F0 | - | - | 0 | 5 | | 0 | • | • | • | |
| -1004 | - | • | ۶ | • | • | ٩ | ٠ | • | • | | ¢ | | ~ | Ģ | Ģ | Q | | | N | |
| DRFRAM | N. | N | ۵ | • | • | 3 | ¢ | • | • | | : 8 | 8 | 66 | 90 | 8 | 5 | | | 99 | |
| FOFESX- | 2 | 2 | • | _ | | 9 | | | | | .25000 | 40880 | 2.66667 | . 12500 | 3.0000 | .14286 | | | 3.66667 | |
| | ٠ | 32 | • | | | 26 | ٩ | | | PELLEO | - | - | 2 | N | e i | N | • | • | , cr | |
| FOFTZOT | | 10 | • | ٩ | ۹ | 2 | ۲ | ۲ | | A-LHN | ۲ | - | 16 | ٩ | 12 | 3 | LO. | 3 | ٩ | |
| 5- | N | 2 | | | | 2 | | | | | | | | | | μ. | | • | ~ | |
| UH-FA- | | | • | | • | - | | • | | | ğ | ÷∰. | 96 | đ | Ř | ġ. | ě. | ¥. | je e | |
| ->L-JA- | CI. | à | | • | • | | • | • | | -XCFEQ | .33333 | 20000 | .61905 | .50000 | -64706 | 3.66667 | . 10000 | 1.57143 | 2.66667 | |
| UORL | • | • | | • | • | | ÷ | • | | | - | | - | - | | | ~ | - | ิณ | |
| | - | | | | | | | | | 0-4 | | | • | • | • | • | • | , | • | |
| UR-HA- | Q | M | • | • | | 10 | • | • | • | | | | | | | g | | | | |
| | - | - | | | | - | • | * | _ | | 00000 | 5000 | | | | . 33333 | | | | |
| じんてらじれる | 4 | 9 | 4 | 0. 0. | 4 | 44 (1) | • | 1 | 00 00 | | | 56 | | | | ŝ | - | | · | |
| 041002- | 0 4 | 6.3 | 4 | 10 | | 6 | | I | 9 | HZGHE- | ~ | N | • | | • | e | | • | . • | |
| | ~ | | , | | | | | 1 | | 医医子 4 | 7 | Þ | N | z | Z | 2 | | | ×- | |
| NCORH | 92 | 22 | 55 | 22 | 40 | 30 | • | Ţ | 4 | COM-ZO4 | 3 | 2 | 4 | • | 9 | 5 | • | • | 0 | |
| 64-19 | ٢ | φ. | - | 9 | \$ | ٩ | • | • | - | FOOF | - | - | 0 | N | ٢ | - | • | • | N ., | |
| RH4 | ٩ | - | 9 | ٩ | | ٩ | • | ÷ | | DZFRAM | - | seid. | 3 | LO. | 3 | 2 | • | • | 3 | |
| re -0 | | ٩ | 9 | , | ٢ | 0 | ۲ | • | ٩ | HOFA <x4< td=""><td>Ì</td><td>1</td><td>4</td><td>9</td><td>ę</td><td>ଧ</td><td>٠</td><td>•</td><td>10</td><td></td></x4<> | Ì | 1 | 4 | 9 | ę | ଧ | ٠ | • | 10 | |
| L ⊂ − Q | 9 | 0 | 3 | • | 9 | 1 | ۲ | 1. | | ZAXDF4 | | 50 | 53 | 0 | Ν | 19 | 0 | • | 21 | |
| | Q | ٢ | 9 | 9 | 9 | | ÷. | | \$ | | | 10 | i) | 112 | 127 | | | | - | |
| 0020 | - | - | - | 9 | | • | · † . | ٩ | - | FOF-ZG4 | e | 2 | ø | 10 | 9 | 16 | • | ۹ | 4 | |
| | | - | - | 1.0 | | 9 | •: | | 9. | 63 mž | | | | | A 1 | | _ | | - - | |
| 0DE0 | 6 | - 0 | 0 0 | 0 | 9 1 | 1 0 | • | | _ | 401~140 | 0 61 | 1.5 | 9 | 2 2 | 4 | 5 | : | | 10 10 | |
| | 6 | 6 | 0 | 6 | 6 | 6 | | с т | | 「「「」」」の | () () | 8 | 3 3 | | е Т | | | | 2 | |
| H-St | 6 | 4 | 6 | 8 | . | N | | j | ~ | CORF4 | - 61 - 61 | 0 | 8 | 6 | | | | | 6 | |
| 409 209 | 0 | 5 | | ~ | 0 | | • | | | ••==• | Ψ. | H | | | 10 | - | | | Ξ. | |
| -KOF | | - | ŝ | - | | 6 | • | | - | CH-FB4 | 0 | 2 | 10 | 6 | Ν | | . • | • | | |
| | - | - | 2 | - | - | ۰ ق | • | • | - | 王下山田4 | 3 | 3 | \$ | C1 | N | ŝ | ٠ | | 0 | |
| 9 | - | - | - | | - | 9 | ٠ | • | - | 3 人団大 | N | N | 1 | N | Z | 1 | | | . ≯ . | |
| 2 - | 10 | 4 | 2 | 2 | 3 | ø | • | • | - | UOZ-zdø | 9 | | 4 | • | _ | Ŷ | • | • | · [• | |
| 90 | 3 | | 2 | ŝ | 0 | ,= | • | • | - | 94004 9404 | | - | - | Ģ | | | • | • | | |
| 7 4 | 50 | 9 | 200 | 400 | 200 | 750 | • | • | 400 | HOFEAND | 4 | 10 4 | 3 | 8 | 6 5 | 5 N | | | 33 | |
| | KJ | 1500 | ลั | 4 | ă | Ň | | | 4 | | - N- | | | | • | | • | | 6.2 | |
| S H E S | 6 | | 6 | 6 | • | 0 | | | 8 | X axale | • | 196 | 37 | 86 | 274 | 2 | 0 | ٩ | 12 | |
| | 750 | 7500 | 400 | 2500 | 750 | 200 | | | 1599 | | | ÷. | | | | | | | | |
| | | | | | | | | | | -ozo | 10 | Ν | 8 | 21 | 18 | 15 | \$ | 9 | T. | |
| 5-1-4 | 756 | 8 | 100 | 2500 | 750 | 2500 | • | • | 2500 | | 5 | C) | 13 | ŝ | 4 | 5 | • | сі — с 1. • | 10 | |
| | M | 15000 | -3- | 25 | 2 | 25 | | | 23 | 02-1-20 | LO. | 4 | 10 | 4 | 5 | 2 | • | | 5 | |
| | - | 5 | 2 | e | 0 | 4 | - | | .00 | E F 140 | 3 | e | ¢ | 3 | 3 | 8 | • | •••• | 3 | |
| AES 1- | 4 | 6 | 3 | 9 | 4 | 2 | | | 10 | UOZLO | 10 | ► | 9 | 10 | 8 | ÷. | • | • | 9 | |
| | 4. 4 | 10 9 | 8 | ý | 41 V | - 10 | - | • | - 10 | UM-Fac | -10 | N | 10 | 9 | 9 | 8 | | | 10 | |
| | • | - H | | • | • | • | | | Т. | SEL-JA | 6 | - | .0 | N | N | | | | 8 | |
| K <f-0< td=""><td>٢</td><td>_</td><td>9</td><td>8</td><td>9</td><td>٩</td><td>•</td><td>•</td><td>0</td><td>M∃≻0</td><td>z</td><td>z</td><td></td><td>7</td><td>Z</td><td>z</td><td>¥</td><td>`</td><td>z</td><td></td></f-0<> | ٢ | _ | 9 | 8 | 9 | ٩ | • | • | 0 | M∃≻0 | z | z | | 7 | Z | z | ¥ | ` | z | |
| RAH0 | 9 | - | 0 | 9 | 9 | 3 | • | • | 9 | COMHZON | 0 | _ | 4 | | - | 0 | 4 | 3 | e | |
| 84F0 | • | | - | | | | ٠ | • | - | HOUHN | - | - | 4 | - | - | 0 | 3 | 8 | - | |
| RAFF | | | 0 | - | - | - | • | • | - | DEFECXO | . ന | | 10 | 2 | цĢ | 3 | 5 | 4 | 3 | |
| HAF0 | - | - | 9 | 1 | 1 | • | • | • | | FOFEAXO | ಳು | 10 | 21 | ŝ | N | Q | 10 | Ν | 8 | |
| 2<1-5 2<1-5 | - | 0 | 0 1 | 6 | | 9 | • | • | _ | N-X0F0 | - | | | <u>, m</u> | | N - | _ | ~ | | |
| E <f4< td=""><td>00</td><td></td><td>9</td><td>6</td><td>0</td><td>9</td><td></td><td>. !</td><td>6</td><td></td><td>•</td><td>120</td><td>44</td><td>33</td><td>250</td><td>27</td><td>3</td><td>33</td><td>26</td><td></td></f4<> | 00 | | 9 | 6 | 0 | 9 | | . ! | 6 | | • | 120 | 44 | 33 | 250 | 27 | 3 | 33 | 26 | |
| 風へ下の | 0 | | 9 | 6 | 6 | 9 | | | 6 | FOF-ZON | 4 | 9 | | e | | <u>(1</u> | - | - | 8 | |
| HAF- | ō | 6 | 6 | 6 | 6 | 6 | • | • | 6 | . <u>₩</u> 1 | 4. | | 34 | •• | 28 | 22 | 2 |] [| - | |
| UWEF | e | | 8 | 4 | | 8 | • | • | | SUN | l0 | 9 | 13 | 5 | 9 | 0 | 0 | 5 | 0 | |
| | 69 | 62 | | 22 | 23 | 22 | 26 | N | 28 | じほー下へる | 4 | 10 | 0 | to . | 9 | b | 3 | 5 | Ν. | |
| NFDOMIE | | | 12 | | - | | | 22 | | ドモレムの | | . 0 | | 3 | 3 | N | N | 0 | 0 | |
| ALDANTE | 0 | | | N | 2 | . 10 | 2 | | 9 | MERCHIPE | 0 | e. | | 2 | 5 | | 9 | 1 | 8 | 1 |

Table 3-5. Master Data Table (continued)

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| | | (| DRI | GIN | IAL | P | AGi | E 18 | S | | | | | | | | | | | | |
|----------------------------------------|------------|---------|-----|-----|-----|--------------|--------|---------|---------|------|------------------------------------------------------------------------------------------------------------------------|------------------|----------|----------|------------------|---------|----------|---------|----------|-----------|--|
| | | (| OF | PO | OR | Ql | JAL | . T' | 1 | | NHTHO | 13 | | • | • | ٠ | - | •. | ٠ | 3 | |
| | • | | | | | | | | | | ¥ K | = | - | •. | • | • | = | ۲ | ۲ | ** | |
| <>0002-2 | . 80 | 0.80 | | | • | 2.75 | | ••• | | | | 168 | 0.500 | | | | | | | -964 | |
| | 8 | 12- | | | | ୍ ମ ଜ | | | | | DHZOKHE | ล่ | | | | • | લં | • | • | فبتبو | |
| 4>0+00F | 0.8 | 0.6 | ٠ | • | • | • | • | • | • | | | 180 | 5.500 | | | | | | | 182 | |
| • • • • | 1.80 | 2.20 | | | • | -25 | | | | | E-zo-z | ~ | ці. | • | ٠ | • | 3.6 | . • | • | ~ | |
| 4202K-0H | | | • | ٩ | • | ø | • | • | • | | DE-Q4-24 | .8 | 2 | • | ٠ | • | N | • | • | ő | |
| <>UFOFEX | 6.5 | 2.6 | • | ٠ | ٠ | 4 | • | ٠ | . • | | FOFS-EL | - | N | ٠ | • | • | N. | • | ····•• . | Į | |
| | 8 | 31 | | | | 29 | | | | | HOH-Edaa | 5 | FT | • | • | • | 9 | • | • | 2 | |
| | .79998 | 0.69231 | | | | <u>66667</u> | | | | | | 38.45 | B. 00 | | | | - 50 | | | -87 | |
| NH | | | • | • | ٠ | | • | • | • • | | > <el<< th=""><th>38</th><th>8</th><th>ĺ.</th><th></th><th></th><th></th><th>÷</th><th></th><th>4</th><th></th></el<<> | 38 | 8 | ĺ. | | | | ÷ | | 4 | |
| H0H-20 | 16 | 20 | 48 | 30 | 52 | 63 | 21 | 11 | 33 | | | 36 | 88 | | | | . 50 | | | 12 | |
| UOZ-ZO | 6 | 4 | ٠ | | | | • | • | | | ZUZZLA | 10. | 6 | • | • | • | 13. | • | • | 10.55 | |
| FOUL | 4 | Ċ | • | | | 0 | • | • | • | | | 8 | 90 | | | | 2 | | | 2 | |
| DZ-GHEAM | • | - | | | | | | | | | | 1.8800 | 8.8700 | | | | 5.0700 | | | 55200 | |
| | | | • | • | | - 17 | | | _ | | | | | • | - | • | ÷•• | | - | | |
| FOFEAM | 1 0 | 13 | • | • | • | 18 | | | - | | | 6299 | 4500 | | | | 4.3300 | | | . 9417 | |
| | | 1 | | | | 8, 1471 | 143 | 727 | .8864 | | MEAZOF | 2.6 | 5.4 | • | • | ٠ | 4.3 | • | ٠ | 3.8 | |
| zont-et- | | | ٠ | • | • | 8.1 | 8.7143 | 12.2727 | 2.8 | * | FOR-FAL | 12 | 11 | • | • | • | • | • | • | • | |
| NUHZ | 6 | | e | | | | 2 | і. П | | 31 | FOR-FRAM | 17 | 21 | ٠ | ٠ | • | 14 | •. | 4 | • | |
| 2445 | • | | • | • | • | • | • | • | | | | | | 9 | 00 | N | | | | 2 | |
| 212 > LIZ | . • | | | ٠ | • | | • | • | • | | | 6.62007 | .61531 | 5.99146 | 7.82405 | 6.62007 | 7.82405 | | | 7.82405 | |
| | | | | | | 5 | | | 115 | | 66144 | 9.6 | 9.6 | 0.9 | 7.8 | 6.6 | 7.8 | • | • | 2. B | |
| D 4 H 4 | • | ٠ | • | | • | 30. | • | • | 18. | | | 20 | 66 | 46 | 01 | 20 | 32 | | | | |
| | | | | | | 1 | | | | | | 6.62007 | 8.92266 | .99146 | .02405 | 6.62007 | 5.29832 | | | .31322 | |
| | | | | | | .81250 | | | .50000 | | Naf-100 | | | 6 | 2 | | | • | • | N | |
| | ٠ | ÷ | . • | • | ÷ | 9.9 | • | | | | | 000 | 000 | 299 | 299 | 000 | 000 | 000 | 00 | 299 | |
| | | | | | | * | | | - | | 4-0-UOM | | 160 | 666 | ·66 | 000 | 000 | 000 | 0.500000 | 999 | |
| ><#R | • | • | • | • | • | 36.924 | • | • | 161-91 | | | 0.680 | 0.360 | 10 0.666 | 0.466 | 0.400 | 0.600 | 0.000 | 0.1 | 0.666 | |
| | | | | | | 36 | | | 16 | | FOFUCE | 21 | Φ | 10 | N | 9 | 2 | • | 10 | 10 | |
| | | | | | | 29 | | | 16 | | | 0.20 | 0.40 | | | | -25 | | | | |
| Xuqxo | • | | | | | 7.2667 | | • | 6.9091 | | | 9 | | • | • | •. | - | • | • | • | |
| | | | | | | | | | 9 | | 0-4-5 | | 0 | ٠ | • : | | D | • | • | • | |
| | | | | | | 47.968 | 89.614 | 106.818 | 357 | | | 000 | 0.307692 | | | | 000 | | | | |
| ><¤ご | • | • | • | • | • | | .6 | 9.9 | 63.357 | | | 1200 | 201 | | | | 1750 | | | | |
| | | | | | | - | | | | | C>U-ZLFE | 1.60000 0.320000 | 0.3 | • | € ¹ . | | 0.875000 | • | • | • | |
| •••••••••••••••••••••••••••••••••••••• | | | | | | 8.4091 | 8.7143 | 12.2727 | 10.2500 | | | 80 | 46 | | | | | | | | |
| 192761 | • | • | ٠ | | • | 8.4 | B. 7 | 2.2 | 9.2 | | | 609 | 1.53846 | | | | 3.50000 | | | | |
| | | | | | | 1 | - | - | Ä | | PELFEF | | 1. | • | • | • | | ٠ | • | • | |
| >44- | | | | | | 50.322 | | | | | | 000 | 162 | | | | 29 | | | | |
| | | | | | | 20 | 1 | | | | | 1400 | 38 | | | | 666 | | | | |
| | | | | | | 0 | | | | | 42025 | 0.340000 | 0.138462 | ٠ | | • | 0.166667 | • | • | • | |
| - NAEM | | | | | | 10.1000 | | | | | <>0+0+-z | | | 00 | 00 | 33 | | 00 | 00 | 00 | |
| | .• | • | • | • | • | 10. | • | • | • | | | 3.2000 | 4.0000 | 16.0000 | 10.0000 | 17.3333 | 15.7500 | 10.5000 | 5.5000 | 11.0000 | |
| | • | • | . – | - | \$ | 9 | • | • : | - | | | 3 | 4 | 16 | 91 | | 15 | | io. | II | |
| SHOONEH | 69 | 82 | 12 | 72 | 23 | 58 | 92 | 22 | 28 | dij. | WEDAWEE | 69 | 0 | 12 | 20 | 62 | 55 | 92 | 22 | 82 | |

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Table 3-5. Master Data Table (continued)

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| DOGHZOH | 1 | .00 | • | • | | | | | 8 | 66667 | .0000 | 66667 | | | | |
| FOURH | ٠ | 2 | • | • | • | ٠ | | | 2.36364 | | 9 | Ň | | ÷ | | |
| びメホル ヘズ エ | ٠ | 3 | ÷ | ٠ | • | • | | | | | | | | | | |
| HOFEKYH | • | | | | | | D-LHO | Ť | | 25 | 9 | 9 | • | | | |
| Z <xal-< td=""><td>6</td><td>1.1</td><td></td><td>•</td><td></td><td>-</td><td></td><td></td><td>-</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td></xal-<> | 6 | 1.1 | | • | | - | | | - | | | | | | | |
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| FOFHZOH | ۵ | Q | | • | | | HELECO | | 2 | 8 | 8 | 66 | | | | |
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| OR-FA- | • | 61 | • | • | • | ٠ | A-LHN | 9 | | | 10 | • | - | | | |
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| Uozk- | • | | • | · • | • | ٠ | | 16969. | 1.88889 | .22222 | .30060 | .33333 | 62500 | | | |
| | | | | | | | TELTEN | 8 | 3 | 8 | 8 | 88 | 2 | | | |
| 02-F8- | | 0 | • | • | • | • | | • | œ. | . 4 | | . 6 | 9 | | | |
| F538- | . • | 2 | • . | ٠ | •. | ٠ | | 0 | | | 3 | - | N | | | |
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| NCORE | • | 9.2 | 30 | 69 | 69 | • | X A A A | | z | X | N | N. | | | | |
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| | • | 9 | ۹ | 9 | 9 | • | DZEEKY | | 10 | 5 | | - | | | | |
| トローの | • | Ť | • | 9 | • | ٠ | | | | | | • | _ | | | |
| トルージ | • | \$ | ٩ | 9 | \$ | | FOFE <x4< td=""><td>•</td><td>11</td><td>•</td><td>-</td><td>9</td><td>•</td><td></td><td></td><td></td></x4<> | • | 11 | • | - | 9 | • | | | |
| | | | õ | Ō | ō | | ZAXAH4 | s, | 10 | æ | - | à | • | | | |
| じりれら | | _ | _ | _ | - | | | | 15 | 63 | 11 | 99 | • | | | 2 |
| | | | | | | | HOH-ZO4 | • | 26 | 33 | 9 | 24 | 9 | | | |
| UDET | • • • | - | 9 | 9 | 9 | • | | | ର | (7) | | 2 | | | | |
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| =-0F | | 6 | | ō | 6 | | UOZL+ | _ | | • | 9 | 10 | • | | | |
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| 5 - C | ٠ | | | - | - | | | | 2 | N | × | 2 | | | | |
| 6 | | ÷. | 10 | 5 | | | COMHZON | • | . 1 1 | 9 | 4 | 9 | ÷ | | | |
| 00 | | ~ | 3 | | N | | 600Ho | ٠ | - | 4 | | - | ٠ | | | 4 |
| | · · • | | | | | • | DXFAdXo | · • | 2 | M | - | 3 | ٠ | | | |
| | • | 88 | 20 | 20 | 58 | • | FOFEAMO | | O. | N | - | | • | | | |
| | | 13 | | | | | | | •• | Ċ, | | | | | | |
| | | | _ | _ | - | | E <xaho< td=""><td></td><td>20</td><td>53</td><td>୍ଷ୍ଯ</td><td>2</td><td>\$</td><td></td><td></td><td>></td></xaho<> | | 20 | 53 | ୍ଷ୍ଯ | 2 | \$ | | | > |
| 8 H H 8 | • | 1599 | 200 | 1500 | 200 | • | | | .01 | | Ψ | - | | | | |
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| | | 1.5 | | | | | COZIO | • | 61 | N | φ. | • | • | | | É |
| 0 L | • | 5 | 2 | 0 | 0 | • | | | | | | | | | | |
| | •. | 2 | N | 9 | 0 | • | CH-FB0 | • | N. | 4 | 5 | 10 | • | | | |
| 244H-0 | | | | ٩ | \$ | • | L L L L L C C | • | 0 | 2 | 3 | 2 | • | | | |
| HAF-0 | • | • | 0 | | 6 | | X 邑 | Z | 7 | N | ¥ | z | N | | | |
| RAFO | | - | • | . 3 | 6 | | COR-ZON | _ | - | 0 | 10 | | - | | | |
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| EAFN | • | | | - | . # | • | FOURN | - | 3 | • | 3 | | | | | |
| R <f 9<="" td=""><td>•</td><td>9</td><td>9</td><td>8</td><td>Ð</td><td>•</td><td>DNFX<xu< td=""><td>LQ.</td><td>to:</td><td>10</td><td>5</td><td>3</td><td>4</td><td></td><td></td><td></td></xu<></td></f> | • | 9 | 9 | 8 | Ð | • | DNFX <xu< td=""><td>LQ.</td><td>to:</td><td>10</td><td>5</td><td>3</td><td>4</td><td></td><td></td><td></td></xu<> | LQ. | to: | 10 | 5 | 3 | 4 | | | |
| RAF 5 | ٠ | - | 0 | 9 | 9 | • | HOFESAN | | φ. | 9 | 5 | 3 | 0 | | an a | |
| K <f4< td=""><td>۰.</td><td></td><td>0</td><td>0</td><td>9</td><td>.•</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td>A Maria Maria</td><td></td></f4<> | ۰. | | 0 | 0 | 9 | .• | | | | | | | | | A Maria Maria | |
| E < F C | • | Ġ | 6 | - | 9 | | Z <x0h0< td=""><td>a.</td><td>26</td><td>80</td><td>44</td><td>41</td><td>36</td><td></td><td></td><td></td></x0h0<> | a. | 26 | 80 | 44 | 41 | 36 | | | |
| EAFO | | 6 | 6 | — | 6 | - | | | | - | - | | | | | |
| | | - | . – | | ~ | • | FOF-ZON | 23 | 21 | 29 | 33 | 4 | 21 | | | |
| 446- | • | 8 | 9 | 9 | 9 | • | 00 | 6 | 6 | | 9 | ٢ | 6 | | | |
| CHEF | • | 4 | 2 | 3 | 0 | • | UZ-FAN | 6 | 2 | 4 | 2 | ÷ | - | | | |
| | 52 | 8 | 81 | 82 | 83 | | | | | | | | | | | |
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Table 3-5. Master Data Table (continued)

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| FOFEAM | ÷ | 26 | ٠ | ٠ | ٩ | • | ENARAH | | ž | | | 2.615 | | |
| | N | | 63 | 8 | 2 | 5 | | | 4 | Ŧ | • | 2 | ٠ | |
| | 5 | 9 | 31 | 29 | 68 | 1965 | FUEFFAL | • | 2 | | · . | به چ | | |
| | 6.6957 | 7.9167 | 6.3313 | 11.9798 | 13.0682 | - | | | | | | | | |
| | 9 | N | 9 | | 13 | Ţ. | FOR-Feer | | 2 | .* | • | • | • | |
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| ><は5 | (**•) | , | • | | ٠ | · | 66794 | | 8. 16052 | 6.29832 | 8.92266 | 5.29832 | | |
| ZAAR | | .• | | ٠ | ÷. | , ; , , | | | 16 | 5 | 6 | 5 | | |
| | | 2 | 9 | 25 | 8 | | | • | | فر | | 5 | • | |
| >4#4 | | 17.846 | 135.110 | 3.067 | 186.879 | - | | | 2 | 2 | 2 | 32 | | |
| | • | N | 12 | 3 | 8 | • | 00L-JOU | | 13. | ą | 13: | 98 | | |
| | | 1 | | | | | | | 7.31322 | 6.29032 | 7.31322 | 5.29832 | | |
| | | 8 | 6.78788 | 33 | 9.50090 | | | | | | | | | |
| XUAX4 | | | 87 | 3.33333 | 8 | | | | 88 | 33 | 29 | 29 | 88 | |
| | ٠ | | N | . 3 | | , | استعده فتجربون | 8 | 88 | 33 | 66 | 66 | 96 | |
| | | | | | | | THOPOOR | 000000 | 0.050000 | 0.133333 | 9.466667 | 8.466667 | . 000000 | |
| | | .500 | 22.617 | . 333 | .473 | | | | | | • | | 6 | |
| ><¥♡ | • | | 2.5 | 2.3 | 3.4 | ٠ | HOLOOK | • | 2 | 0 | Þ | M | | |
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| | | 7.0000 | 3.9643 | 90 | 4.4545 | | | | N | - | - | | | |
| E B <ec< td=""><td>÷</td><td>Ö</td><td>6</td><td>ē.</td><td></td><td></td><td>A-(</td><td>•</td><td>II</td><td>.</td><td>3</td><td>•</td><td></td><td></td></ec<> | ÷ | Ö | 6 | ē. | | | A-(| • | II | . | 3 | • | | |
| | | · • | 3 | 21.0000 76 | . 🖤 | | | | | | | | | |
| ><12 | | . Q | N | 9 | Ŀ | N | | | 0.519231 | | | | | |
| | 51.858 | | 1 | N | 282.917 | 83.162 | 4>D-RAHR | | 613 | | | | | |
| and a state of the | Ы. | ្តត្ត | 39. | 8 | 32 | 23 | | • | 1.1 | ٠ | ٠ | • | i in | |
| | | 10.5000 322.133 | 8.2414 289.547 | 11.6061 588.746 | 2 | | | | ~ | | | | | , 1997 - 1997 1997 - 1997 - 1997 |
| KU <ko< td=""><td>6.6957</td><td></td><td>. 🛣</td><td>19</td><td>25.23.79</td><td>30</td><td></td><td></td><td>2.07692</td><td></td><td></td><td></td><td></td><td></td></ko<> | 6.6957 | | . 🛣 | 19 | 25.23.79 | 30 | | | 2.07692 | | | | | |
| | 69 | ě | 2 | 69 | 3 | 11.1985 | HELFEF | | 67 | | | | | |
| | 6. | | 8 | | | - | | ÷ | N | • | ۴ | | | |
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| | | 136.167 | ÷ | - | | | | | 46 | | | | | |
| ><#- | • | 1 | : ÷ | : , . ● | | • | 4>0NH | | 63 | | | | | |
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| | | 292 | | | | | | 000 | 906 | 163 | 333 | 900 | 900 | |
| KU <k-< td=""><td>· · .</td><td>. 1667</td><td>· .</td><td>4</td><td></td><td></td><td><>UFOFFE</td><td>11.5990</td><td>13.3009 0.163462</td><td>39.6667</td><td>14.3333</td><td>13.0000</td><td>21.0000</td><td></td></k-<> | · · . | . 1667 | · . | 4 | | | <>UFOFFE | 11.5990 | 13.3009 0.163462 | 39.6667 | 14.3333 | 13.0000 | 21.0000 | |
| | | | • | - | | | | 11. | 13 | 6 | 1 | | 21 | |
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| OFDANEF | 5 | 8 | 6 | 8 | | 3 | NFDUNKF | 62 | 8 | • | ď | Ö. | 3 | |
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Table 3-5. Master Data Table (continued)

MAXIMUN 03000000 "I 00000000 "I 00900009 "I 00900009 "I 00900000 "I 00000000 "I 00000000 "I MINIMUM I.12500000 I.2500000 .00000000 1428
 14.09332254

 1.43617161

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 200.23750307

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 2025503625

 2.023750436

 2.023750436

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Table 3-6. STATISTICS FOR PERFORMANCE MEASURES

-66-

(continued)

Table 3-6.

| MAXI MUM VALUE | 26.4444444 28.333333 21.0000000 13.0000000 15.7751000 15.777775 638.730333 638.700000 | |
|-----------------------|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|--|
| MINIMUM | 3.0000000 5.333333333333333333333333333333333333 | |
| STANDARD DEVIATION | 5.73186168 6.28365104 3.96574182 2.95669341 2.73333221 121.48373560 121.48373560 157.52487389 156.12591031 | |
| MEAN | $\begin{array}{c} 10.62921245\\ 12.14518399\\ 8.40835212\\ 7.36637337\\ 9.66473353\\ 9.66473353\\ 102.85220992\\ 232.70653447\\ 26.05824707\\ 96.28053447\\ 96.28052575\\ 96.28052575\end{array}$ | |
| K, | 86986868688 | |
| VARIABLE | HEAN I HEAN2 HEAN3 HEAN4 HEAN4 MDELTAT VAN3 VAN3 VAN3 VAN3 | |

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list in order to allow all forty-three subjects' data to be considered in the regression analysis. Since the majority of subjects completed scenarios 2, 3, and 4, all calculations for the independent performance measures were limited to these three. Given the criteria just mentioned above, only knowledge score and total correctness (TOTCORR = C2 + C3 + C4) can be appropriately modeled.

When correctness scores on individual scenarios are examined, C1-C4, several interesting conclusions appear. Knowledge subscores are good predictors for C2 and C4 only. A single experience variable, source of instrument training, shows up in C3 prediction. The total number of unique tracks and the total number of times the subject was on the correct track are positive predictors in either scenarios 1 or 2. The lower DIFTT (total tracks - total unique tracks), the higher the correctness score is on scenario 1. The correctness score on scenario 3 increases as the total number of inquiries decreases. As might be expected, the lower the variance on delta (decision and response) time, the higher the correctness score on scenario number 4. What was surprising was the absence of such occurrences over all four correctness scores. This may be due in part because good performance on one scenario does not necessarily translate into good performance on all four scenarios as verified by the learning experiment.

For Z_t the ratio of total correct to total number of tracks, it was revealed that as both DIFTT (TOTRAK - UNIQTRAK) and the total number of inquiries diminish, Z_t rises. A pilot's source of instrument training seems to augment Z_t . Both Z_t and the total number of inquiries are negative predictors of mean time between inquiries (MDELTAT).

Table 3-7. SUMMARY OF STEPWISE PREDICTIONS

| | PREDICTOR VARIABLE | | BIOGRAPHICAL & EXPERIENCE | KNOWLEDGE | INDEPENDENT Performance vars. |
|--------|-----------------------|----|------------------------------|-------------------------------------|---------------------------------------------------------------------------------------------------------------------|
| | R ² | N | | | |
| | | | SET TT SO AGE | CATSCR1 SCORE CATSCR2 CATSCR3 | CL-C4, INPTRT, DIFTT, MEAN1-4, VAR1-4, TOTINQ, TOTRAK, UNIQTRAK, CORINQ, TOCT MDELTAT, TOTCOR, ET, VDELTAT |
| Cl | .65 | 26 | | | C2 (.0031)* UNIQTRAK (.0001) DIFTT (.0847)* C3 (.0072)* C4 (.0600)* |
| C2 | .80 | 26 | | CATSCR2 (.0527)* CATSCR3 (.0011) | TOCT (.0014) C4 (.0024)* VAR1 (.0011) MEAN3 (.0001)* |
| C3 | .93 | 26 | SO (.0001)* | | Cl (.0536)* UNIQTRAK (.0001) TOTINQ (.0021)* MEAN1 (.0169) VAR4 (.0162)* |
| C4 | .66 | 26 | | CATSCR3 (.0044) | C2 (.0066)* MEAN 1 (.0139) MEAN 2 (.0103)* VAR 4 (.0366)* |
| ZT | .79 | 26 | SO (.0144) | CATSCR1 (.0550)* | DIFTT (.0696)* TOTINQ (.0259)* MEAN2 (.0195) VAR2 (.0739) |
| DELTAT | .64 | 26 | SO (.0122) Age (.0357) | | C3 (.0041) ZT (.0016)* TOTINQ (.0091)* |

*NEGATIVE B VALUE

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Referring to Table 3-8 and recalling the procedure utilized in performing this set of regression analyses, certain intuitive results appear. The total number of unique tracks, the total number of times the subject is on the correct track, and the number of single engine hours are all positive predictors of pilot's knowledge survey score. The total number of tracks and knowledge score are inversely related.

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As the total number of tracks and the total number of times the subject is on the correct track increase, total correctness (TOTCORR) will also increase. Total correctness will rise as DIFTT (total tracks - total unique tracks) decreases.

Tests On Data Partitions

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Table 3-9 labeled "Summary of T-Test" shows a series of tests on extreme partitions of major independent and key dependent measures to determine if differences might exist between them. These extreme "cuts" were obtained from a cumulative frequency table furnished by SAS. The splits or cuts were made at the lower 25 percent quartile and at the upper 75 percent quartile on biographical measures involving flying time and age, plus two performance measures, SCORE and TOTCOR. For example, the cumulative frequency table reveals that for this group of subjects the lower 25 percent have, at most, only one hundred hours of single engine flying experience, while the upper seventy-five percent have greater than 1,001 hours of single engine experience. Partitions were also made according to pilot classifications, i.e. private versus non-private, VFR rated versus IFR rated, military versus civilian, high recency versus low recency, and pleasure versue non-pleasure. -70-

Table 3-8. SUMMARY OF STEPWISE PREDICTIONS

| | REDICTOR VARIABLE | | BIOGRAPHICAL & EXPERIENCE | KNOWLEDGE | INDEPENDENT PERFORMANCE VARS. |
|---------------------------------------|----------------------|----|------------------------------|------------------|--------------------------------------------------------------------|
| · · · · · · · · · · · · · · · · · · · | R ² | N | | | |
| SCORE | ,51 | 35 | SETLOG (.0934) | | TOTRAK (.0009)* UNIQTRAK (.0109) C3 (.0776)* TOCT (.0078) |
| TOTCORR | .48 | 35 | | CATSCR1 (.0191)* | TOTRAK (.0934) DIFTT (.0125)* Toct (.0004) |

-based on Scen 2,3,4

*NEGATIVE B VALUE

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Table 3-9 Summary of T-test

| | Test Variables | | | | |
|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|----------------|--------------------------------------------------------------------------------------------------------------------------------------------------------------|---------|---------------------------|------------------------------------|
| Data partitions on Biographical & Experience and Key Dependent Measures | N | Variable | Prob> T | Higher Valued Group | N for spec. varia ble |
| | | CATSCR1, CATSCR2, CATSCR3, SCORE, C1-C4, TOTC TOTRAK, UNIQTRAK, DIFTT, TOCT, CORINQ, TOTINQ INPTRT, SETLOG, TTLOG, MEAN1-4, MPELTAT, VAR1-4, SO, ZT | | | |
| Low Single Engine Hrs. vs. High Single Engine Hrs. | | CATSCR2 | (.0361) | (1) | 6 10 |
| (0) ≤ 100 | (0) 10 | CATS CR3 | (.0478) | (1) | 6 10 |
| (1) ≥ 1001 | (1) 11 | SCORE | (.0148) | (1) | 6 10 |
| | | SETLOG | (.0001) | (1) | 6 11 |
| | | TTLOG | (.0001) | (1) | 6 11 |
| | | MEAN2 | (.0365) | (1) | 10 9 |
| | | VAR2 | (.0391) | (1) | 10 9 |
| Low Total Hrs. vs. High Total Hrs. | | CATS CR2 | (.0176) | (1) | 15 8 |
| (0) ≤ 300 | (0) 19 | CATS CR3 | (.0958) | (1) | 15 8 |
| (1) ≥ 2001 | (1) 9 | SCORE | (.0624) | (1) | 15 8 |
| | · · · · · | Cl | (.0367) | (0) | 11 6 |
| | | SETLOG | (.0001) | (1) | 15 9 |
| | | TTLOG | (.0001) | (1) | 15 9 |
| Private vs. Non-Private Rating | | CATSCR2 | (.0090) | (1) | 16 21 |
| (0) Private | (0) 16 | CATS CR3 | (.0995) | (1) | 16 21 |
| (1) Commercial & Air | (1) 22 | SCORE | (.0334) | (1) | 16 21 |
| Transport | | TOTRAK | (.0761) | (0) | 13 16 12 |
| | | UNIQTRAK | (.0745) | (0) | 13 16 |
| | | SETLOG | (.0001) | (1) | 16 22 |
| nen el ser el presentação de ser el 1990 en 19 A ser el ser A ser el ser e | | TTLOG | (.0001) | (1) | 16 22 |

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| Table 3-9. (continued) | | ORIGINAL PAGE IS OF POOR QUALITY | | | |
|-------------------------------------------------|---------|-------------------------------------|---------|-----|----------------|
| Young vs. Old | | CATSCRI | (.0393) | (1) | 3 5 |
| (0) < 20 | (0) 7 | SCORE | (.0557) | (1) | 3 |
| (1) ≥ 51 | (1) 5 | | | | |
| IFR vs. VFR | | CATS CR2 | (.0587) | (0) | 20 17 |
| (0) IFR | (0) 21 | CATS CR3 | (.0787) | (0) | 20 17 |
| (1) VFR | (1) 21 | SCORE | (.0582) | (0) | 20 17 |
| | | TOTCOR | (.0447) | (0) | 21 20 |
| | | SETLOG | (.0015) | (0) | 21 17 |
| | | TTLOG | (.0001) | (0) | 21 17 |
| | | VAR4 | (.0987) | (1) | 16 17 |
| | | SO | (.0488) | (1) | 21 17 |
| | | PTOTCOR | (.0178) | (0) | 21 20 |
| Low Knowledge Score vs. High Knowledge Score | | CATSCRL | (.0005) | (1) | 9 10 9 |
| (0) ≤ 45 | (0) 14 | CATS CR2 | (.0002) | (1) | 10 9 |
| (1) > 65 | (1) 1.0 | CATS CR3 | (.0001) | (1) | 10 14 |
| | | PTOTCOR | (.0585) | (1) | 14 10 14 |
| | | C2 | (.0700) | (1) | 10 14 |
| | | TOTCOR | (.0324) | (1) | 10 |
| | | SETLOG | (.0072) | (1) | 10 10 |
| | | TTLOG | (.0183) | (1) | 10 10 7 |
| | | MEAN1 | (.0816) | (1) | 7 |
| | | MDELTAT | (.0227) | (1) | 13 7 |

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| Table 3-9. (co | ntinued) | ORIGINAL FAGE IS OF POOR QUALITY | | | |
|------------------------------------------------------|----------|-------------------------------------|------------------------------------------|-----|----------|
| Civilian vs. Military | | | | | |
| (O) Civilian | (0) 32 | C2 | (.0674) | (0) | 31 6 |
| (1) Military | (1) 6 | TOCT | (.0149) | (0) | 24 5 |
| | | TTLOG | (.0021) | (1) | 32 6 |
| | 1 | VAR1 | (.0427) | (0) | 24 4 |
| | | VAR3 | (.0100) | (0) | 27 5 |
| Low Total Correct vs. High Total Correct | | | | | 7 |
| (0) ≤ 6 | (0) 11 | CATS CR2 | (.0768) | (1) | 9 3 |
| (1) > 12 | (1) 9 | C1 | (.0279) | (1) | 9 |
| | | C2 | (.0453) | (1) | 10 9 |
| | | C3 | (.0013) | (1) | 6 |
| | | C4 | (.0134) | (1) | 6 9 |
| | | MEAN1 | (.0150) | (1) | 4 , 8 |
| | | MDELTAT | (.0629) | (1) | 10 8 |
| | | ZT | (.0219) | (1) | 3 9 |
| Low Percentage Correct vs High Percentage Correct | • | | an a | | |
| (0) ≤ .35 | (0) 11 | C2 | . (.0282) | (1) | 10 10 |
| (1) > .60 | (1) 10 | C4 | (.0145) | (1) | 7 10 |
| | | | | | 6 |
| | | CORINQ | (.0660) | (1) | 8 |
| | | MEAN1 | (.0058) | (1) | 6 7 |
| | | VAR1 | (.0987) | (1) | 6 7 |
| | | ZT | (.0153) | (1) | 5 8 |

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Table 3-9. (continued)

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| High Recency vs. Low Recency | | | | | |
|---------------------------------|-------------|----------|---------------------------------------------------------|-------------------------------|-----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| (0) Last 30 days | (0) 23 | CATSCR2 | (.0403) | (0) | 18 10 |
| | | | | | 18 |
| (1) > 2 years | (1) 10 | CATS CR3 | (.0092) | (0) | 10 18 |
| | | SCORE | (.0017) | (0) | 10 |
| | | C3 | (.0252) | (0) | 18 9 |
| | | | | | 14 |
| | | TOTRAK | (.0147) | (1) | 8 14 |
| | | DIFTT | (.0100) | (1) | 8 |
| | | TOTINQ | (.0017) | (1) | 23 10 |
| | | IOLINQ | | | 19 |
| | | SETLOG | (.0171) | (0) | 10 |
| | | TTLOG | (.0277) | (0) | 19 10 |
| | | | | | 19 |
| | · · · · · · | MEAN2 | (.0580) | (0) | 10 14 |
| | | ZT | (.0661) | (0) | 7 |
| Pleasure vs. Non-pleasure | | | erten auf der Aller des gerauten der Aller Handlich | | <mark>ta ta produktion de la competizione de la Competizione de la competizione de la compe</mark> |
| (0) Airline | | | ************************************ | | |
| | (0) 17 | CATS CR3 | (.0576) | (0) | 17 20 |
| Military | (0) 1/ | CATSURS | (.0578) | (0) | 17 |
| Hired | (1) 21 | SCORE | (.0471) | (0) | 20 15 |
| Business | | C1 | (.0241) | (1) | 14 |
| (1) Discours | | SETLOG | (.0125) | (0) | 17 21 |
| (1) Pleasure | | SETLOG | · (•0783) | | 17 |
| | | TTLOG | (.0004) | (0) | 21 |
| | | L | <u>an dia mandri di</u> di ka | <u>ala di sana di sana di</u> | |

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Thirty-one performance and experience variables were chosen as candidates for the t-test. The objective was to see if there was any significant difference between the means of the twentyfive percent quartile and the seventy-five percent quartile for the performance or experience measure in question.

Separating single engine experience revealed a higher knowledge survey score for pilots with over 1,001 hours versus those with less than one hundred hours. Total flying experience (\leq 300 hours versus \geq 2,001 hours) appeared to exhibit the same relationship to performance as did single engine flying experience. Splits on ratings revealed that commercial and ATP pilots tend to possess greater knowledge than do private pilots. Private pilots tend to use a larger number of total tracks and total unique tracks than do commercial or ATP pilots.

For training partitions, military training led to smaller variances on delta times (decision and response time) for scenarios 1 and 3. Civilian pilots tend to have fewer hours than do pilots with military training. However, civilian pilots attained a higher correctness score on scenario 2 (vacuum pump failure). The total number of times the subject is on the correct track is also higher for the civilian pilots.

A split on IFR versus VFR rated pilots disclosed that IFR airmen retain a higher knowledge survey score and greater total correctness than do VFR airmen.

When performance measures are split to get profiles of high score pilots versus low score pilots, some independent performance effects are noted. Higher knowledge scores (>65%) are associated with:

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- (a) higher single engine hours experience
- (b) higher total flying experience
- (c) higher correctness scores
- (d) higher mean delta times

When the top and bottom quartile on correctness scores are examined, added independent performance results begin to appear. High total correctness (>12; >60%) is characterized by the following:

- (a) higher knowledge scores on cockpit instrumentation and electrical systems
- (b) a greater number of inquiries on the correct track
- (c) higher mean delta times

The e pilots who submitted an instrument flight rules plan in the last thirty days are characterized by:

(a) higher knowledge scores

- (b) a higher correctness score on scenario 3
- (c) a higher ratio of total correct to total number of tracks (Z_{+})
- (d) greater total flying experience
- (•) fewer total tracks
- (f) fewer DIFTT
- (g) fewer total inquiries

as opposed to those who have not done so in the past two years.

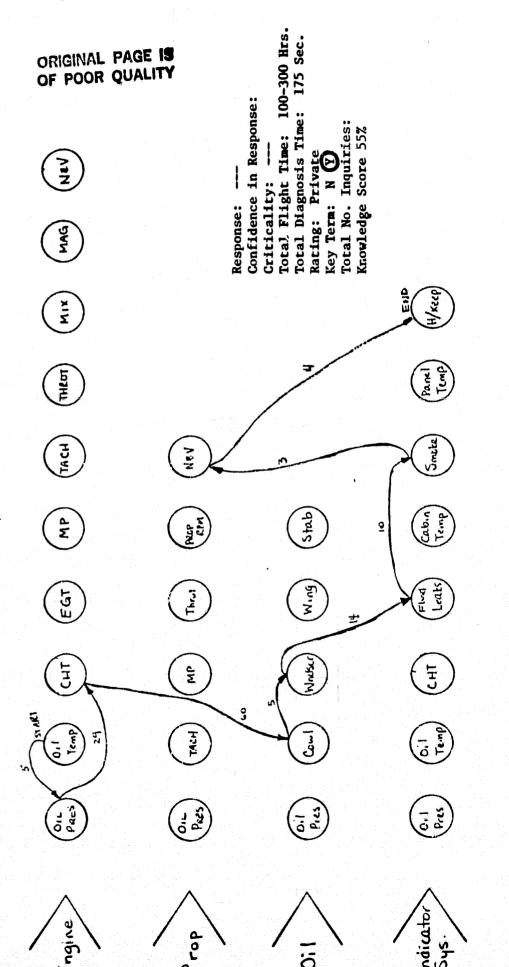
Pilots who fly mainly for pleasure tended to score better on scenario 1. However, this class of pilots showed less knowledge and less experience than people who fly mainly for airline, commercial, business, or military purposes. Airmen over fifty years of age achieved better knowledge scores than those under twenty years of age.

PIP Analysis

A Pilot Information Plot (PIP) was created for every subject on each diagnostic scenario attempted. These charts serve much the same purpose as the flow process charts long in vogue by industrial engineers. They provide a quick visualization of the information gathering style of a subject. One can tell at a glance which subjects have a systematic approach to collecting information and which seem to "shotgun" across the panel in search of any clue to the problem at hand. For example, one can see extremes in subject performance on scenario 2 from the PIPs presented and discussed in chapter 2 of this report.

It is also possible to compare performance of a single subject across all scenarios. For example, subject 42 is depicted in Figure 3-13 through 3-16. This subject has a very systematic search pattern which is similar for all four scenarios. He sought multiple data on a single track before switching. In three of the four he found the key element of information but in each case he elected to seek additional information to verify what at that point must have been an almost certain conclusion.

Subject 57 on the other hand exhibits an almost random data acquisition pattern. Figure 3-17 through 3-20 show evidence of many changes in tracks and a great number of inquiries. It appears as if this subject has no hypotheses to test but rather collects all the data possible in the time allotted before drawing any conclusions. This subject also returns to tracks and even to items within track many times. Yet in spite of this total immersion in data the subject did correctly diagnose two of the four scenarios taken simply by accidently discovering the key information element.



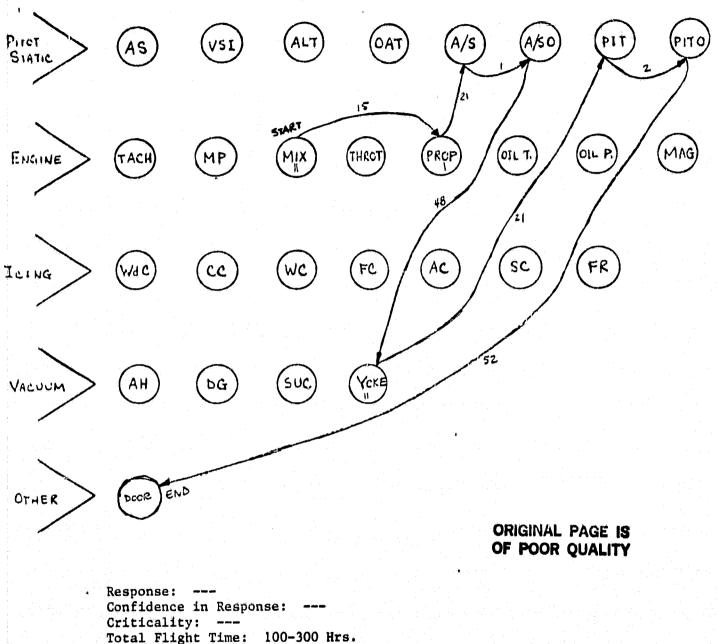
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}

Other

Figure 3-14: Pilot Information Plot For Scenario #2, Subject #42

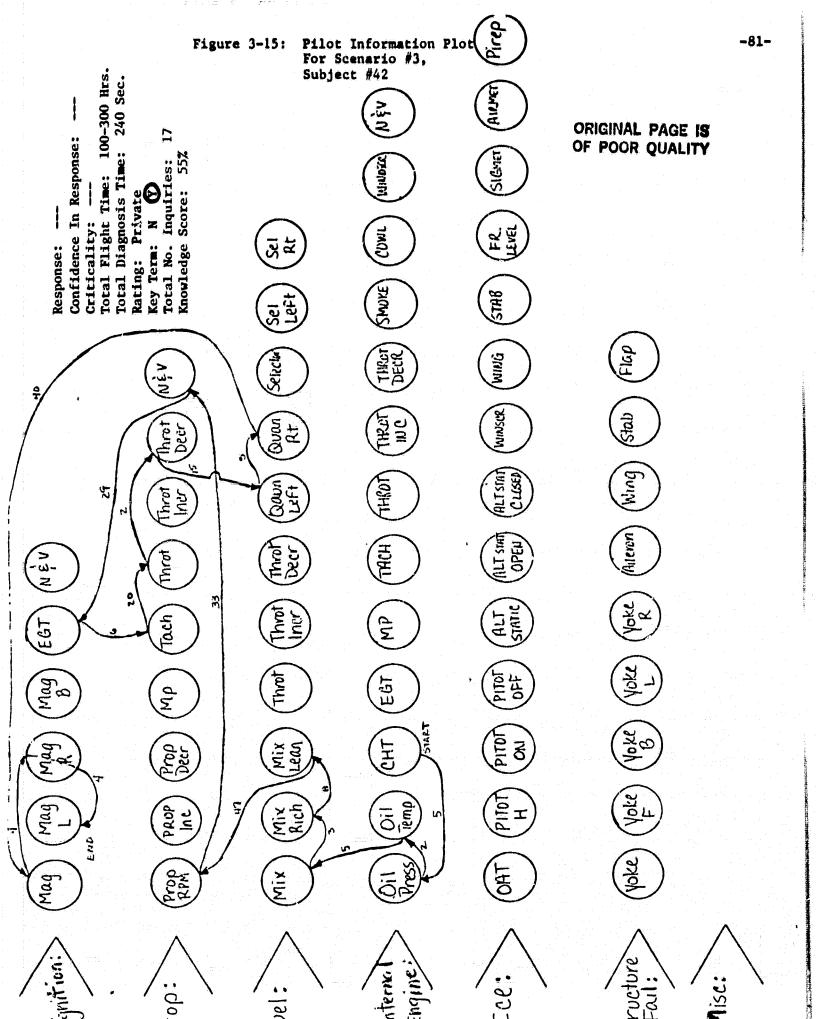
110.022...

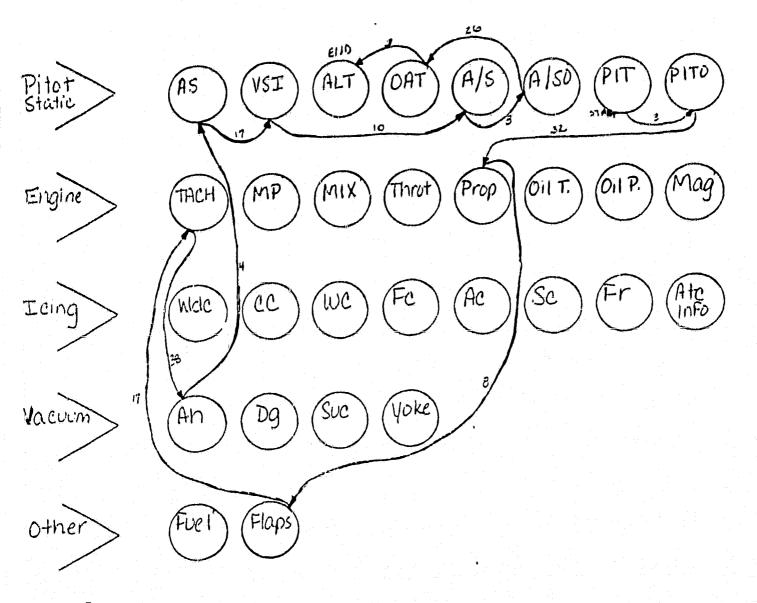


Total Flight Time: 100-300 Hrs. Total Diagnosis Time: 241 Sec. Rating: Private Key Term: N Y Total No. Inquiries: 13

Knowledge Score: 55%

-80-





Response: ---Confidence in Response: ---Criticality: ---Total Flight Time: 100-300 Hours Total Diagnosis Time: 185 Sec. Rating: Private Key Term: N Y Total No. Inquiries: 13 Knowledge Score: 55%

Figure 3-16: Pilot Information Plot For Scenario #4, Subject #42

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301-500 Hrs. Total Diagnosis Time: 145 Sec. 9 Response: Oil Line Broken Original page is of poor quality £ **Confidence In Response:** 50% NIN Total No. Inquiries: Total Flight Time: Fotal Local Rating: Private Knowledge Score: ~ MAG Criticality: E VKEEP MIX Pa.nel Tamp THEOT Smult **FACH** ٧٧٧ Tenn Stab ЧP ģ 6u.W Lead Y Ther EGT Witter CHT ЧP CHT (O. l Temp) 5 **T:**9.6H ini Dic. 1 Dies Pres dw Pacs Press 2 Indicator Sys. Other Engine Prop

Figure 3-17: Pilot Information For Scenario #1, Subject #57

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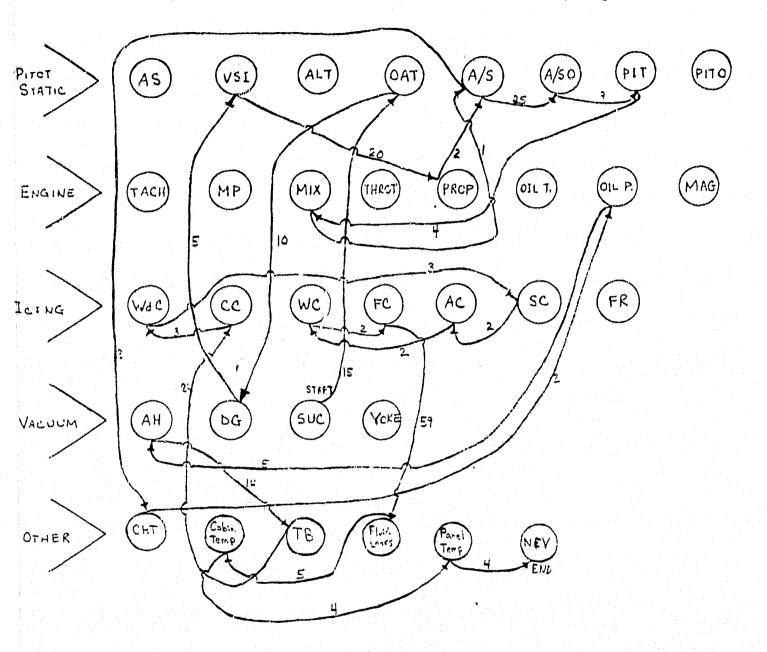


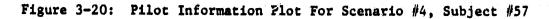
Figure 3-18: Pilot Information Plot For Scenario #2, Subject #57

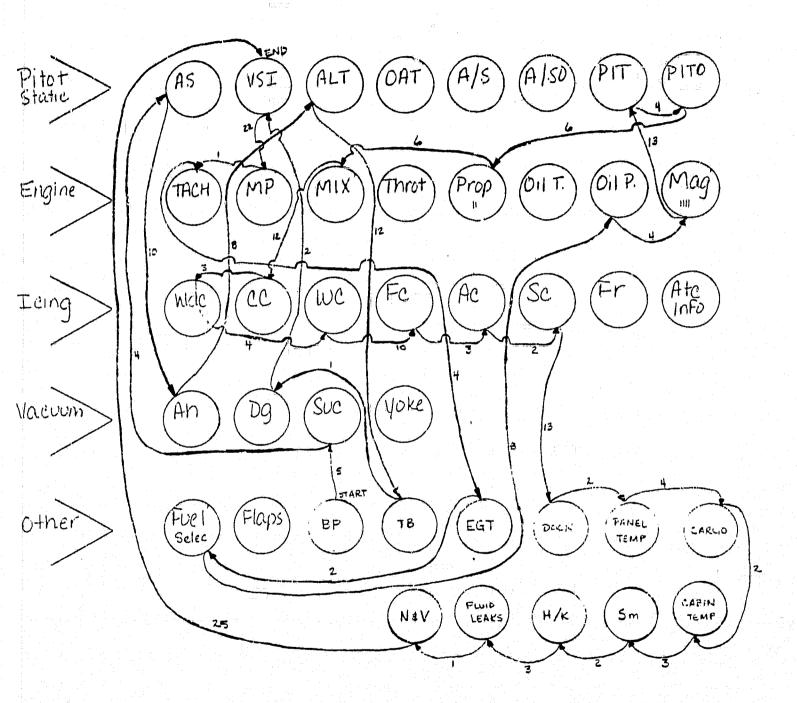
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Response: Vacuum Pump Failure Confidence in Response: 5 Criticality: 4 Total Flight Time: 301-500 Hours Total Diagnosis Time: 241 Sec. Rating: Private Key Term: N Y Total No. Inquiries: 25 Knowledge Score: 50%

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| Figur | e 3-19: | Pilot Information I Scenario #3, Subject | Plot For st #57 | Pirep | ORIGINAL PAG | -85- E IS |
|-----------------------------------------------------------------------------|--------------------------------------|---------------------------------------------|--------------------|--------------|--------------------------------------------------------------------|-------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| oken se: 6 301-500 Hrs. : 240 Sec. | | | N. S. | ALEMEN | OF POOR QUA | |
| n Br Sspon Sspon | Y ries: 36 :: 50% | | END | SIGMET | | |
| ise: Pisto lence in Re ality: 7 Fight Tim Diagnosis | ra: N Y No. Inquiri Ige Score: | Sel | TIMOJ | HER. | | |
| Response: P Confidence i Criticality: Total Fiight Total Diagno | Key Term: Total No. Knowledge | Sel | SHOKE | STHB | t | |
| | Niv | Selection | DECR | String | Flap | |
| | Ihret Na ar | Burn (Rt | THEAT | Munisce | Stat | Corres Corres |
| | Thret | (Down) | LIOCHL | GLISSE | e (Ming | Contraction of the second |
|) - 3- N | Throt | Decr | HIHCH | (ALT SIM | Raileron | Cherui |
| EGT | Tach | Threat | dW | ALT | Noke Ske | LEAKS |
| Mag | dW |) (Internet | | PITOT 0FF | L L | t (and) |
| Mag | Prep | - Lean " | CHT | PITO | - Acke | EN |
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| Mag | (Prop | | | (tes | (P) (P) (P) (P) (P) (P) (P) (P) (P) (P) | A CONTRACT OF A |
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Response: Structural Ice Confidence in Response: 7 Criticality: 6 Total Flight Time: 301-500 Hrs. Total Diagnosis Time: 241 Sec. Rating: Private Key Term: N Y Total No. Inquiries: 40 Knowledge Score: 50% -86-

Schema Analysis

For purposes of comparison, schema diagrams were prepared for two groups of subjects for each diagnostic scenario. The "correct" subjects were those who scored the maximum of five points on a given scenario while the "incorrect" ones scored zero. Each diagram presents the percentage of subjects who included the indicated information in their search versus the median percent order in which that data was requested. (See Figures 3-21 through 3-29).

The schema diagrams for correct and incorrect groups in scenario 1 (Figures 3-22 and 3-23) look remarkably similar. This scenario involves a broken oil pressure gauge line behind the instrument panel. Both groups show a high percentage of interest in oil pressure, oil temperature, and cylinder head temperature (cluster 1) very early in the search process. They also have a similar interest in external evidence of oil on the cowling and windscreen with roughly half of the subjects making such inquiries (cluster 2). General engine health as evidenced by cluster 3 seems to be of less interest. The only distinguishing characteristics appear to be that the correct group looked for fluid leaks, which was the key element, and was concerned with inside physical evidence such as housekeeping and cabin temperature (cluster 4) while the incorrect group ignored this information.

Scenario 2 (Figures 3-24 and 3-25) shows more variation between groups. The only common element shows up in cluster 2 which concerns information related to icing conditions. Roughly half of all subjects seem to believe that this problem, a vacuum pump failure, is ice related. The successful group does check cluster 1, the gyro instruments and suction gauge, while very few in the unsuccessful group bother to look. Cluster 3 in the correct group represents a small minority of inquiries concerning general aircraft health. The incorrect group does not evidence any strong clustering tendency beyond their concern for ice.

The distinction between correct and incorrect groups in scenario 3 appears to hinge on "cluster fixation" (Figures 3-26 and 3-27). Both groups show a timely concern for general engine health, cluster 2, but only the correct group breaks out of the general health symptoms to check the engine ignition system. This cluster (#1) of course contains the key element since the symptoms are caused by a broken right magneto drive gear.

Scenario 4 involving a frozen static port is somewhat perplexing in view of the great interest in icing symptoms evidenced throughout all the scenarios (Figures 3-28 and 3-29). Although both groups do show strong interest in the potential for ice,(cluster #1) the incorrect group seeks that information early and then abandons that track before reaching the key element. The correct group shows a much broader concern and keeps hammering at the ice question until they uncover the static system problem, albeit rather late in the search process.

Figure 3-21. LIST OF SYMBOLS USED IN SCHEMATA.

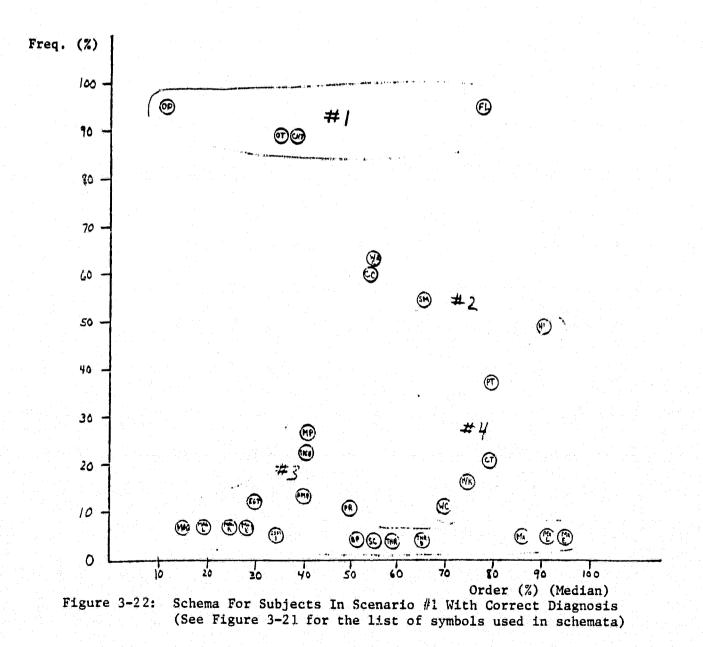
Abbreviations

Aileron Cond = ACAirmets = AirAlt. Static = A/SAlt. Static Closed = A/SCAlt. Static Open = A/SOBreaker Panel = BP Cabin Temp = CT Cargo = Car Ceiling = Ceil Cowling Condition = CC Cloud Tops = CLTCycle Head Temp = CHT Flap Condition = FC Flaps = FFluid Leaks = FLFreezing Level = Fr Fuel Selec = FSFuel Selec. Left = FSL Fuel Selec. Right = FSR Ground Speed = GrS Left Fuel Quant. = LFQ Housekeeping = H/KMag Both # Mag B Mag Left = Mag L Mag Right = Mag R Master Switch = MS Master Switch On = MSOMaster Switch Off = MSOF

Mix = MxMix Enrich = MxEMix Lean = MxLNav. Aid Stat. = NAV Noise & Vibration = NV Oil Pressure = OPOil Temp = OTPanel Temp = PT Pireps = PirPitot Heat = PH Pitot Heat Off = PHOF Pitot Heat On = PHO Prop RPM = PRProp PDM Decr = PRDProp RPM Inc = PRI Right Fuel Qty = RFQ Signets = Sig Smoke = Sm Stabilizer Cond = SC Suction = Suc Throttle = Thr Throttle Dec = ThrD Throttle Incr = ThrI Winds Aloft = WA Windscreen Cond = WdCWind Cord = WC Visibility = Vis

Subjects: 42, 43, 46, 47, 51, 52, 53, 54, 55, 56, 57, 59, 60, 61, 64, 67, 68, 80

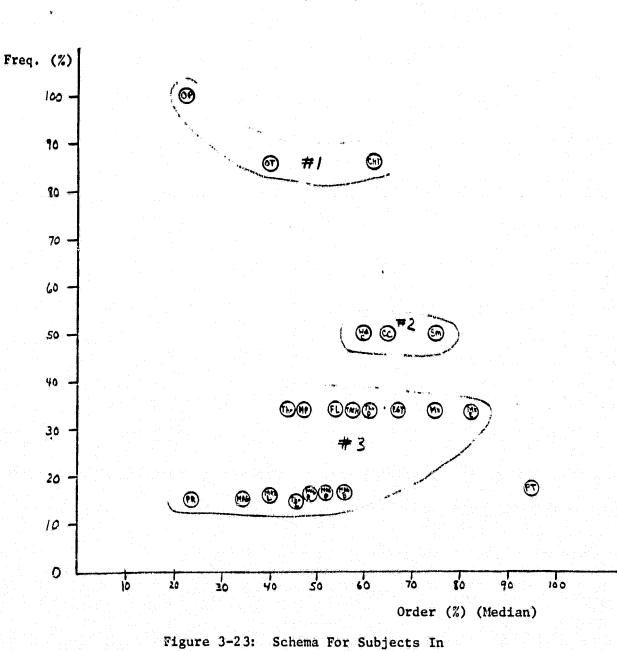
Frequency (%) vs. Order (Median)



Subjects: 45, 48, 49, 50, 58, 65

Frequency (%) vs. Order (Median)

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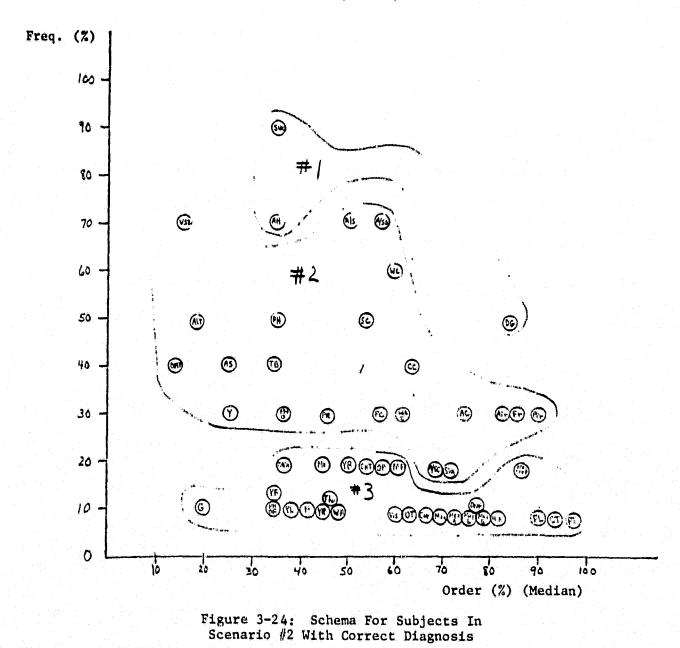


Scenario #1 With Incorrect Diagnosis

-92-

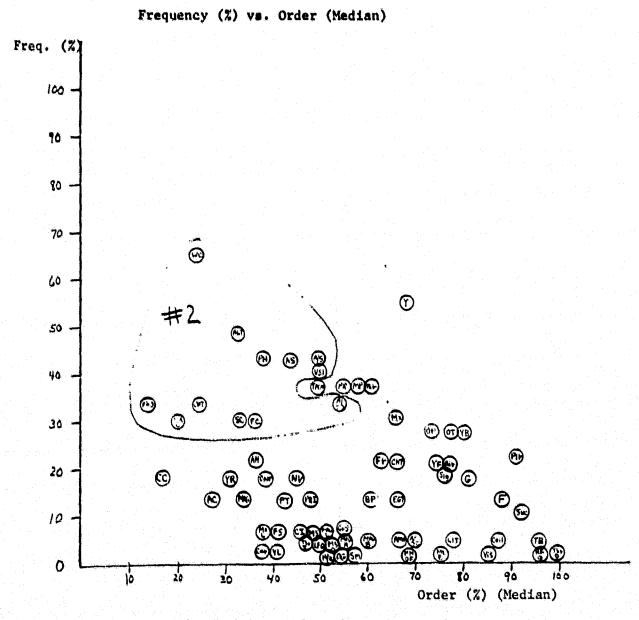
Subjects: 45, 49, 50, 51, 57, 63, 71, 72, 77, 80

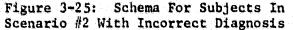
Frequency (%) vs. Order (Median)



Subjects: 42, 44, 46, 48, 52, 54, 55, 56, 58, 59, 60, 62, 64, 65, 66 66, 68, 70, 73, 75, 76, 78, 79, 81, 82, 83, 84

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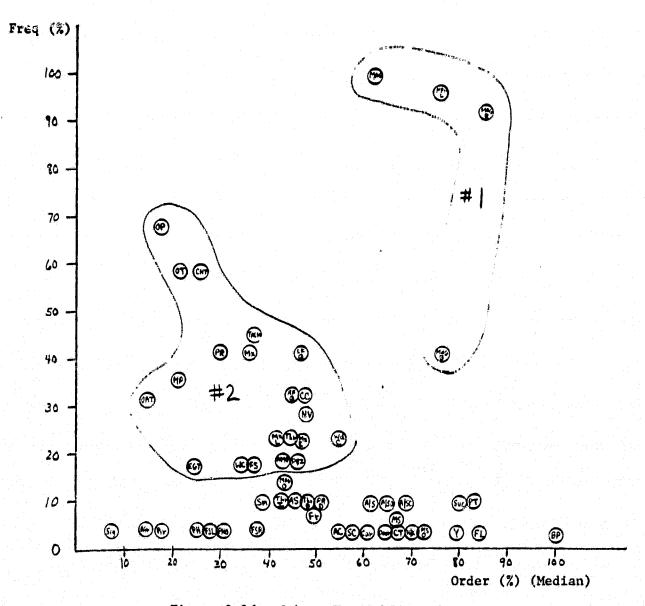


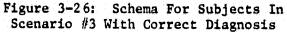
-93-

A

Subjects: 42, 43, 46, 49, 50, 51, 52, 53, 54, 56, 58, 59, 60, 63, 65, 71, 73, 75, 78, 80, 82, 83

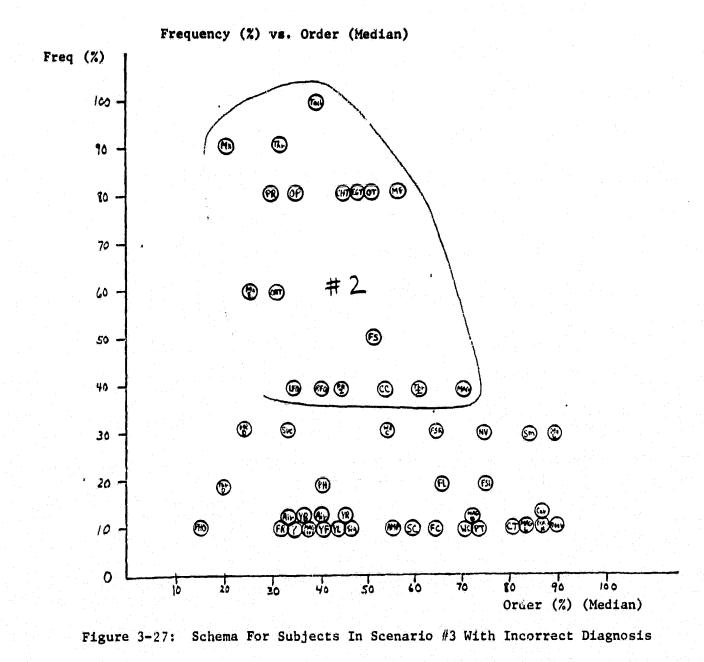
Frequency (%) vs. Order (Median)





-94-

Subjects: 45, 47, 48, 55, 56, 61, 62, 66, 68, 72



Subjects: 42, 43, 46, 48, 50, 51, 53, 55, 58, 61, 64, 65, 70, 75, 78

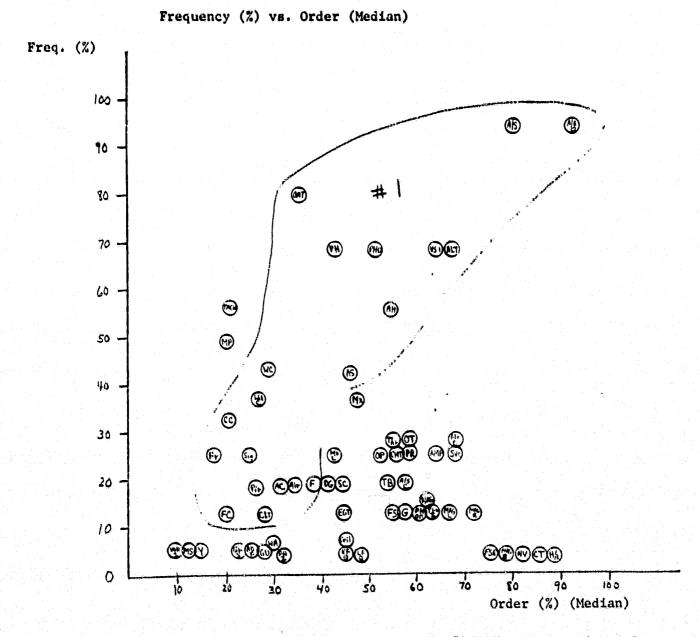


Figure 3-28: Schema For Subjects In Scenario #4 With Correct Diagnosis

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Subjects: 45, 47, 49, 54, 71



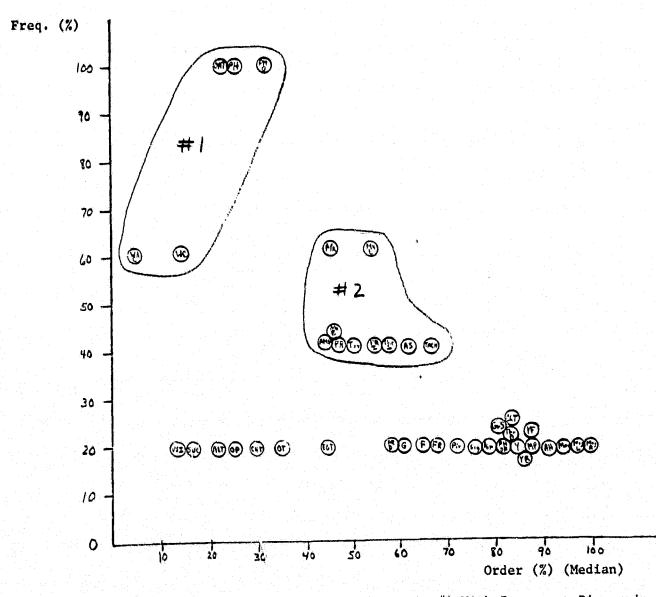


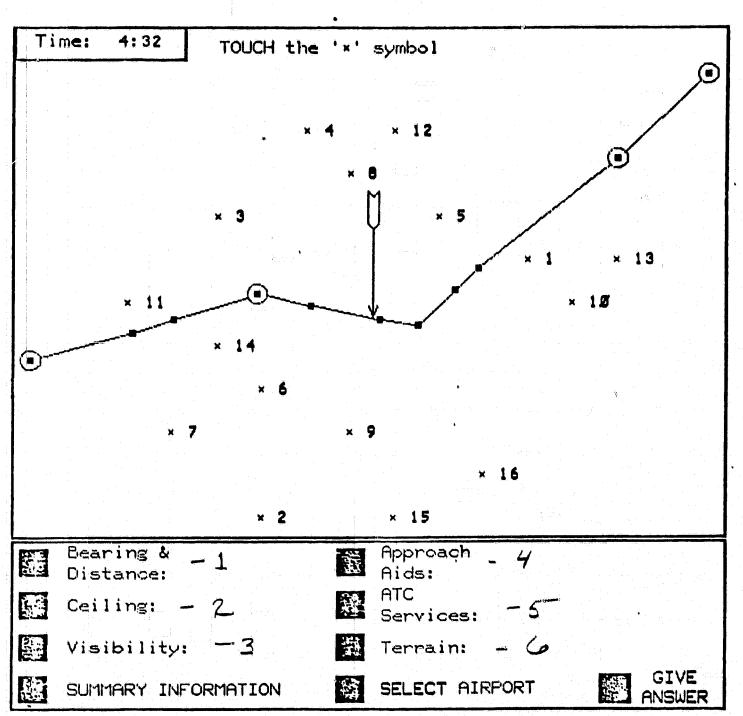
Figure 3-29: Schema For Subjects In Scenario #4 With Incorrect Diagnosis

G. Destimation Diversion Performance

The destination diversion problem involved selecting an airport from among sixteen candidates distributed around the geographic area in which the alternator failure occurred. Six items of information could be requested for each airport during the two minute search period. The information available included: (1) bearing and distance, (2) ceiling, (3) visibility, (4) approach aids, (5) ATC services, and (6) terrain. The location of the airports relative to the point of failure is depicted in Figure 3-30. The problem is somewhat complicated by the presence of a strong southwest wind which was noted in the presentation of the scenario.

The characteristics of each of the sixteen airports are noted in Table 3-10. The characteristics were chosen to represent extremes in desirability, e.g. mountainous versus level terrain, ILS versus NDB approaches, 500 foot versus 1000 foot ceilings, etc.

The number of information requests among the thirty subjects who completed this part of the experiment is summarized in Table 3-11. The items of information most often requested were ceiling and visibility which seems consistent with the order of information in weather reports. The low number of requests for terrain information (rank 5 of the 6 classes) is somewhat surprising. One wonders about the wisdom of selecting an airport on the basis of electronic aids which may face imminent failure without knowing the potential hostility of surrounding terrain. The airports receiving the most information requests were airports 3 and 5. The high frequency of requests for those airports may be due simply to their perceived proximity to the point of failure as determined from the simplified



-99-

| AIRPORT | BEARI DISTA | | CEILING | VISIBILITY | APPROACH AIDS | ATC Services | TERRAIN |
|---------|------------------|----|---------|----------------|------------------|-----------------|---------|
| 1 | 080 ⁰ | 60 | 500 | 2 | ILS | TWR | LEVEL |
| 2 | 230 ⁰ | 50 | 700 | 1 (1) | ILS | NONE | HILLY |
| 3 | 330 ⁰ | 60 | 1000 | 3 | VOR | TWR(R) | HILLY |
| 4 | 350 ⁰ | 65 | 500 | 1 | NDB | NONE | HILLY |
| 5 | 060 ⁰ | 35 | 700 | 2 | NDB | NONE | MOUNT |
| 6 | 270 ⁰ | 40 | 700 | 2 | NDB | NONE | MOUNT |
| 7 | 270 ⁰ | 50 | 500 | 2 | ILS | NONE | MOUNT |
| 8 | 010 ⁰ | 50 | 700 | 2 | NDB | NONE | MOUNT |
| 9 | 200 ⁰ | 25 | 500 | 1 | NDB | FSS | HILLY |
| 10 | 100 ⁰ | 53 | 500 | 1 | ILS | TWR(R) | LEVEL |
| 11 | 290 ⁰ | 65 | 700 | 2 | ILS | NONE | MOUNT |
| 12 | 030 ⁰ | 60 | 1000 | 3 | ILS | NONE | MOUNT |
| 13 | 090 ⁰ | 70 | 1000 | 2 | ILS | TWR(R) | LEVEL |
| 14 | 290 ⁰ | 35 | 700 | 2 | VOR | TWR | MOUNT |
| 15 | 190 ⁰ | 40 | 700 | 2 | ILS | TWR | HILLY |
| 16 | 150 ⁰ | 40 | 500 | 1 | VOR | TWR(R) | LEVEL |
| | | | | | | | |

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Table 3-10: AIRPORT CHARACTERISTICS

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| | Info | Bear Dist | . Ceiling | Visibility | Approach Aids | ATC Services | Terrain | | Frequency of Selection |
|-------|-----------|--------------|-----------|------------|------------------|-------------------------|---------|-------|---------------------------|
| | Airport \ | 1 | 2 | 3 | 4 | 5 | 66 | Total | |
| | 1 | 9 | 15 | 15 | 12 | 10 | 11 | 72 | 8 |
| | 2 | 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 |
| 1 | 3 | 16 | 18 | 18 | 14 | 13 | 11 | 90 | 10 |
| | 4 | 0 | 4 | 3 | 1 | 0 | 1 | 9 | . 0 |
| | 5 | 11 | 21 | 20 | 16 | 13 | 11 | 92 | 1 |
| | 6 | 12 | 16 | 15 | 14 | 8 | 7 | 72 | 2 |
| | 7 | 2 | 2 | 2 | 2 | 0 | 0. | 8 | 1 |
| | 8 | 11 | 16 | 14 | 11 | 9 | 9 | 70 | 0 |
| | 9 | 11 | 18 | 17 | 9 | 5 | 5 | 65 | 0 |
| | 10 | 3 | 6 | 6 | 4 | 2 | 5 | 26 | 0 |
| | 11 | 3 | 6 | 5 | 3 | 1 | 1 | 19 | 0 |
| | 12 | 2 | 4 | 4 | 3 | 4 | 4 | 21 | 2 |
| | 13 | 0 | 1 | 1 | 1 | 0 | 1 | 4 | 0 |
| | 14 | 11 | 20 | 17 | 12 | . 9 ¹ | 11 | 80 | 3 |
| | 15 | 0 | 0 | 0 | 0 | 1 | '0 | 1 | 0 |
| | 16 | 2 | 4 | 4 | 3 | 2 | 3 | 18 | 1 |
| Tota: | L | 93 | 151 | 141 | 105 | 77 | 80 | 647 | ²⁸ /30 |

* Two subjects out of 30 did not select an airport

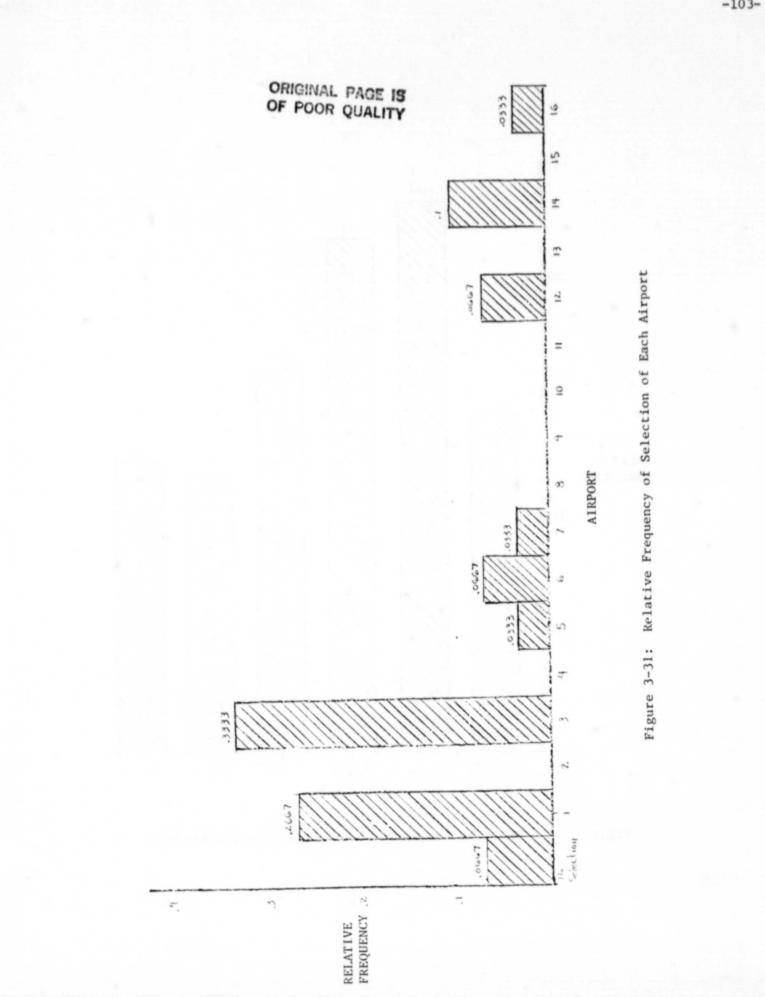
Table 3-11 Destination Diversion Information Requests and Decisions

enroute chart. However, had the subjects considered wind effects, airport 3 should have received much less attention than some others.

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The most popular choices for destination diversion airports were airports 1 and 3. On the surface these two represent distinctly different choices. Both are equidistant from the point of failure. However, airport 1 is in level terrain, downwind and has an ILS approach, while airport 3 is in hilly terrain, upwind with a non-precision VOR approach. Ceiling and visibility are 500 and 1 for airport 1 versus 1000 and 3 for airport 3. Several t-tests were run to test the effect of flying experience, knowledge test scores, diagnostic scenario performance and airport information type requests on the choice of one of these two airports. None of these proved to be statistically significant. Whatever influenced the choice of airport 1 versus airport 3, it is not immediately apparent from the subject characteristics examined.

Figures 3-31 through 3-33 present relative frequency information derived from the table of information requests. Here it is possible to see at a glance the dominance of airports 1 and 3 and information on ceiling and visibility. It is interesting to note that the information frequency profile for those two airports is virtually identical. Profiles for other airports exhibit decidedly different patterns. -102-



-103-

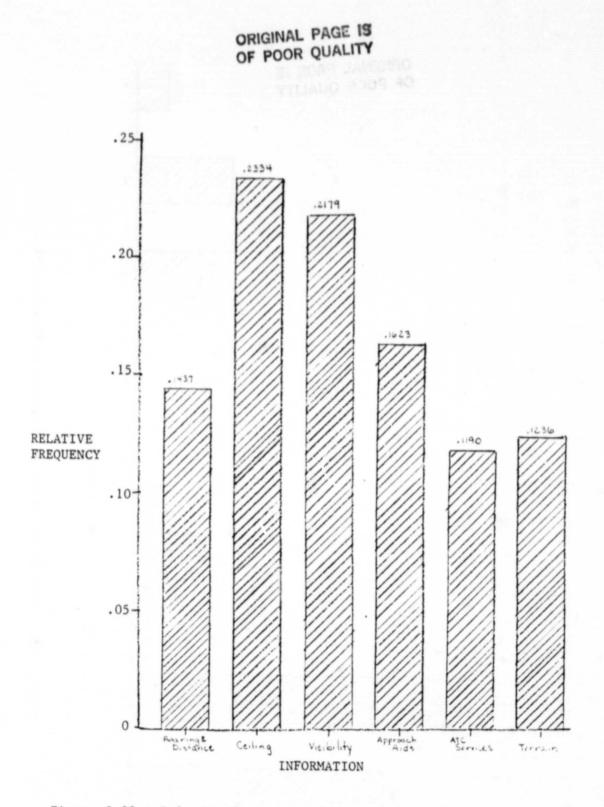
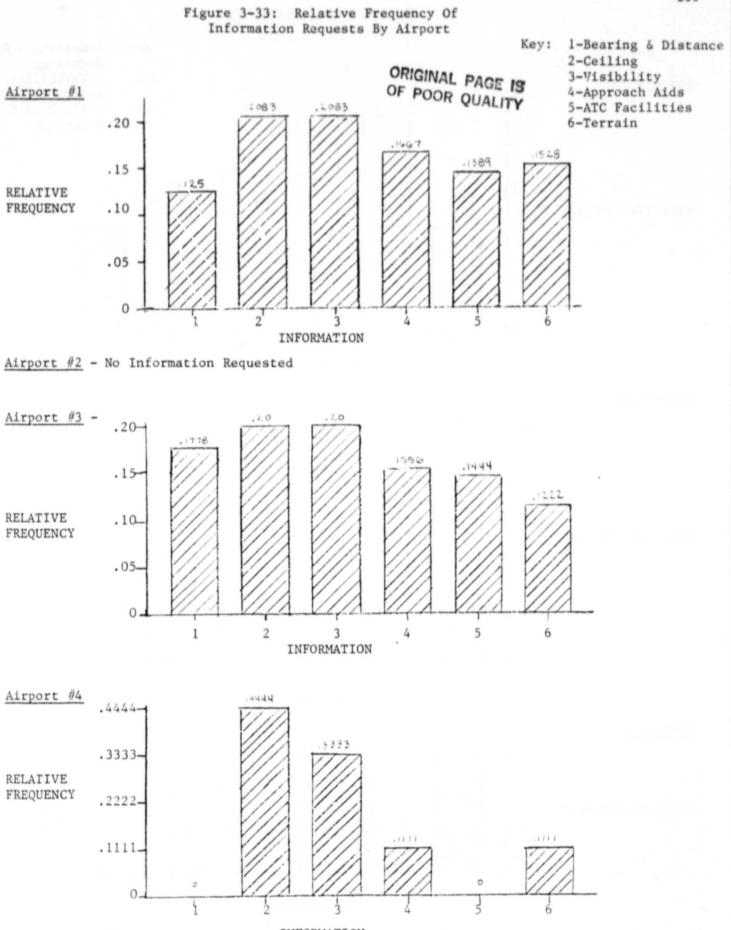


Figure 3-32: Relative Frequency of Total Information Requests



INFORMATION

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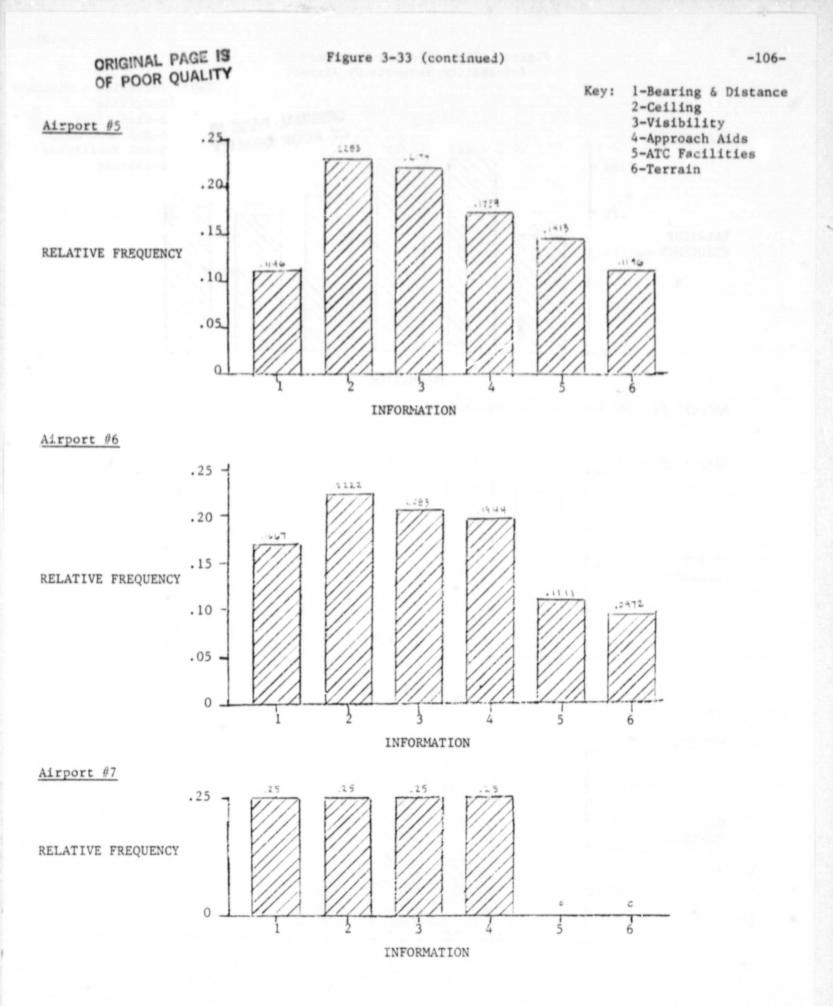


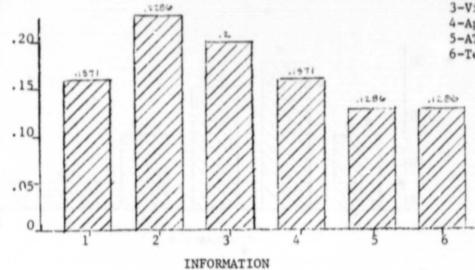
Figure 3-33 (continued)

Key: 1-Bearing & Distance 2-Ceiling 3-Visibility 4-Approach Aids 5-ATC Facilties 6-Terrain

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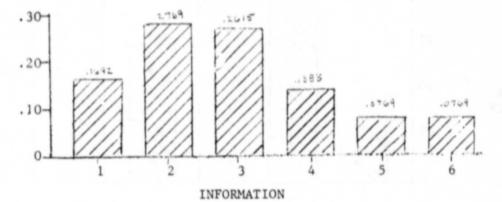
RELATIVE FREQUENCY

Airport #8

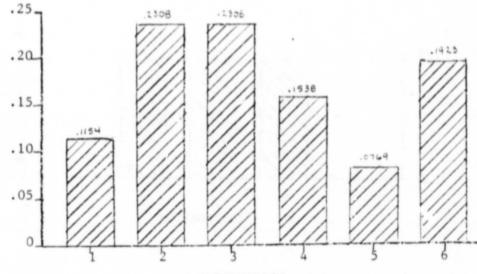


Airport #9

RELATIVE FREQUENCY



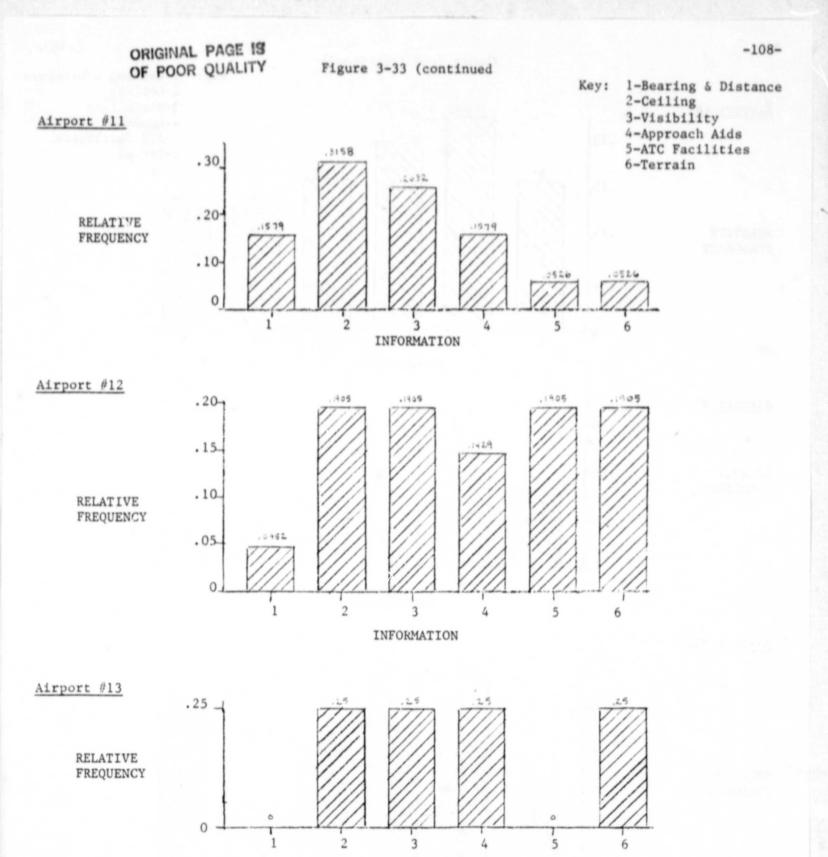
Airport #10



INFORMATION

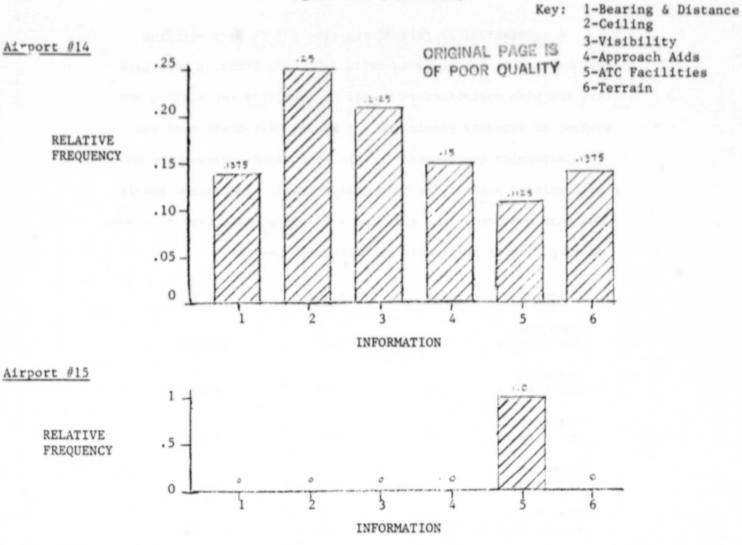
RELATIVE FREQUENCY

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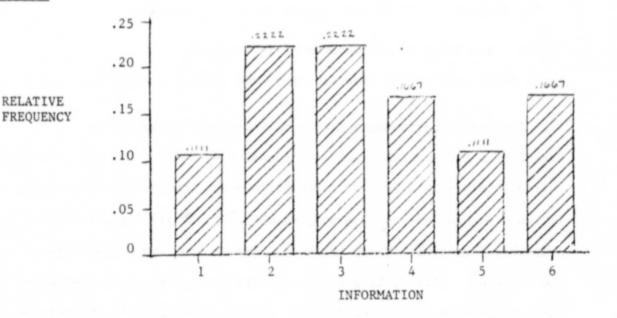


INFORMATION





Airport #16



-109-

A statistical profile of the time and inquiry patterns of subjects is contained in Table 3-12. The items of note here are the wide variations in number of inquiries per subject and number of airports considered. A series of t-tests were run to distinguish the behavior of the professional versus the nonprofessional pilot. The only statistically significant result was that non-pro pilots required a longer average time to choose an airport once their data gathering was complete.

| ITEM | MEAN | RANGE | VARIANCE | | |
|----------------------------|------------|-------------|----------|--|--|
| Inquiries Fer Subject | 22 | 6-44 | 84.8571 | | |
| Airports Par Subject | 6.28 | 2-15 | 9,2069 | | |
| Information Per Airport | 3.82 | 2-6 | 1.4292 | | |
| Time Per Inquiry | 3.44 sec. | 2.15-5.88 | 1.1442 | | |
| Time To Choose Airport | 14.02 sec. | 6-26.5 | 21.7067 | | |
| Time Per Airport | 27.39 sec. | 13.11-52.33 | 89.2388 | | |

Nota: Data based on performance of 29 subjects.

Table 3-12. Destination Diversion Time and Inquiry Statistics

Individual performance differences are best noted in the destination information graphs. Figures 3-34, 3-35, 3-36, and 3-37 depict four subjects with different information needs and information seeking styles. Figure 3-34 shows a subject who collects all the information available from each airport selected. Figure 3-35 depicts a subject who has concern for ceiling and visibility but not much consciousness about terrain and other attributes. Figure 3-36 depicts a subject who

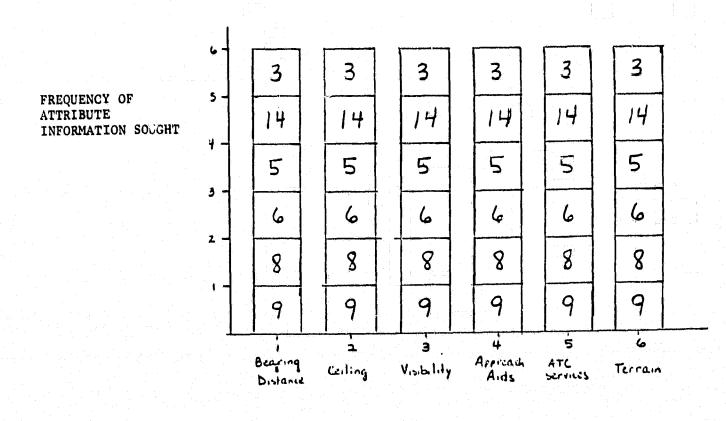
-110-

chooses an airport as soon as it meets his criteria and fails to use available time to locate "better" diversion airports. Figure 3-37 depicts a subject who first faceses on approach aids and only seeks added data on those meeting his approach aid criteria. The secondary information he seeks includes ceiling and visibility.

These DIGs are merely representative of the general styles of information search exhibited. Although each subject can be categorized into one such pattern, there is no discernible relationship between the way subjects collect information for diagnosis as illustrated by PIPs and the way they collect information in destination diversion decisions as depicted by DIGs.

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Subject #51 Time Lapsed: 174 Sec./Port Selection: 10.625 Sec./Inquiry: 2.78 Airport Selected: #3 Note Nos. in the box refer to a specific airport



ATTRIBUTE INFORMATION

Figure 3-34: Destination-Diversion Total Information Search

Subject #78 Time Lapsed: 189 sec. Sec./Port Selection: 10.55 Sec./Inquiry: 3.0417 Airport Selected: #3 Note Nos. in the box refer to a specific airport

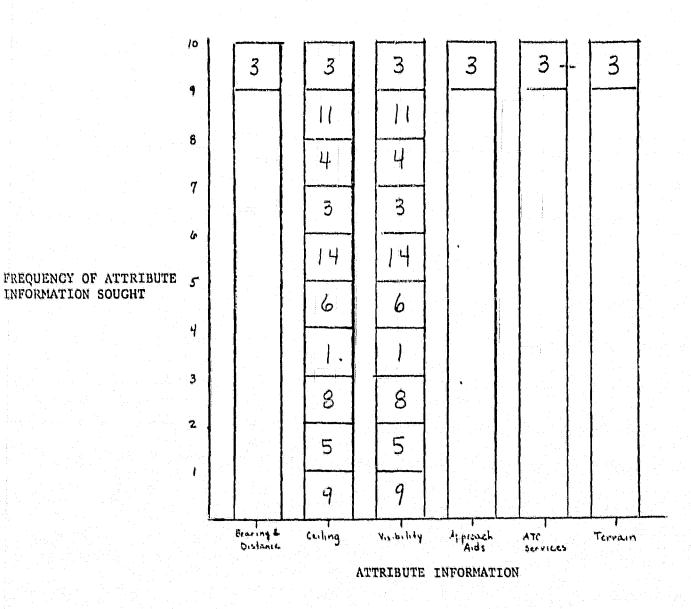
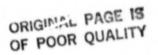
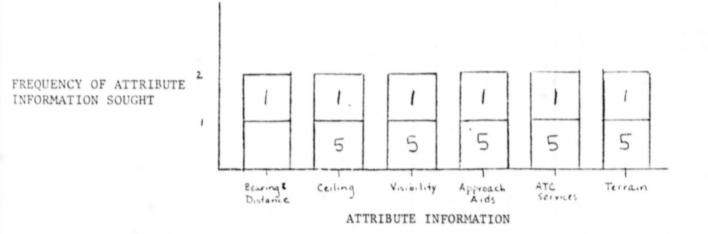


Figure 3-35: Destination Diversion Limited Information Focus

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Subject #70 Time Lapsed: 78 sec. Sec./Port Selection: 9 Sec./Inquiry: 5.45 Airport Selected: #1 Note Nos. in the box refer to a specific airport





-114-

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Subject: #54 Time Lapsed: 129 sec. Sec./Port Selection: 9.625 Sec./Inquiry: 3.25 Airport Selected: #3 Note Nos. in the box refer to a specific airport

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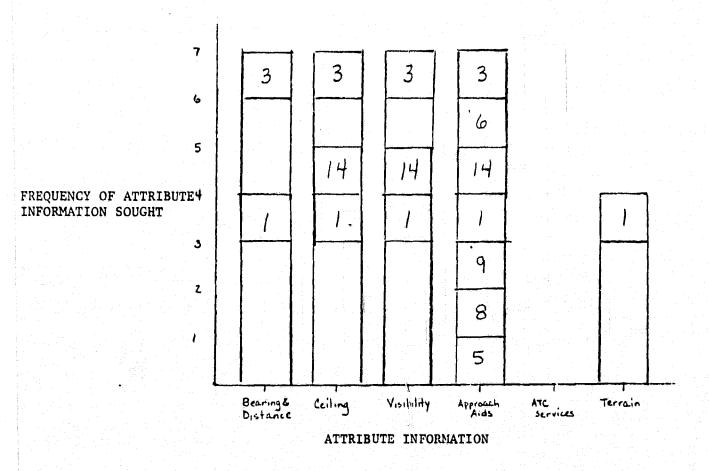


Figure 3-37: Destination Diversion Primary Criterion Search

H. <u>An Experiment Combining Diagnosis and Destination Diversion</u> Decisions Within Computer Aided Testing

Task 6 of the research plan called for an experiment combining the diagnostic and destination decisions within the computer aided tests. This required a simulated full mission scenario which embodied not only a CIFE but also navigational and communication inputs from the subject pilot. This necessitated rather sophisticated programs for the PLATO® system. For navigational responses the aircraft had to be located in space using VOR fixes and the computer had to account for heading and altitude changes imparted by the pilot through an auto pilot display. The navigational and autopilot display is shown in Figure 3-38. The pilot was given a simulated low altitude chart to locate his position relative to VOR's and available diversion airports (see Figure 3-39).

Communication using CAT is difficult because it limits the number of messages the pilot can communicate if one wishes to keep a menu driven display system and not require complex typed responses by the subject pilot.

Communication from ATC to the pilot is less a problem because alpha numeric messages to the pilot are relatively easy to present given that the computer recognizes the pilot's position within the flight. Figure 3-40 depicts the means by which the pilot communicates with ATC. This display is more involved than the ATC information display used in the simple diagnostic tests. The display permits the pilot to ask for information and also to give position reports and declare emergencies or request special handling.

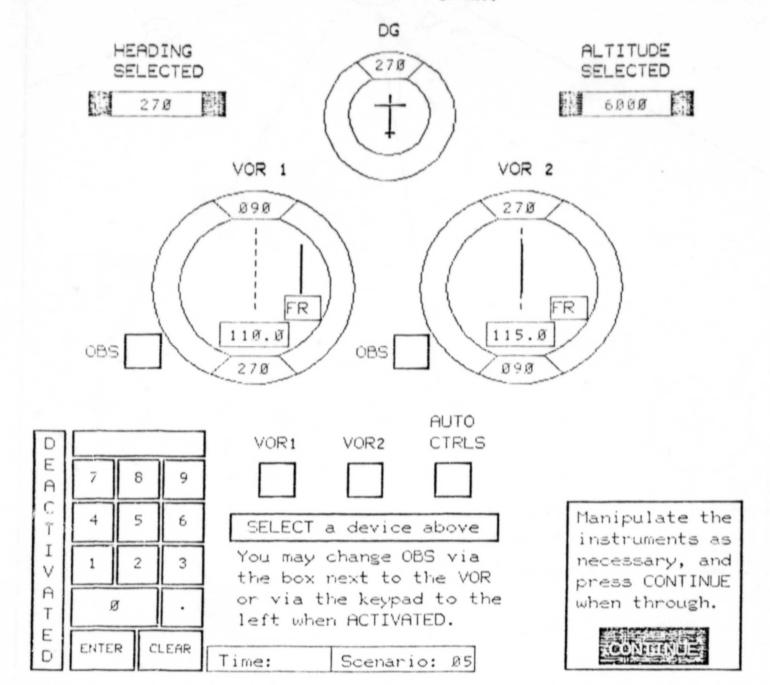


Figure 3-38: VOR-AUTO PILOT DISPLAY

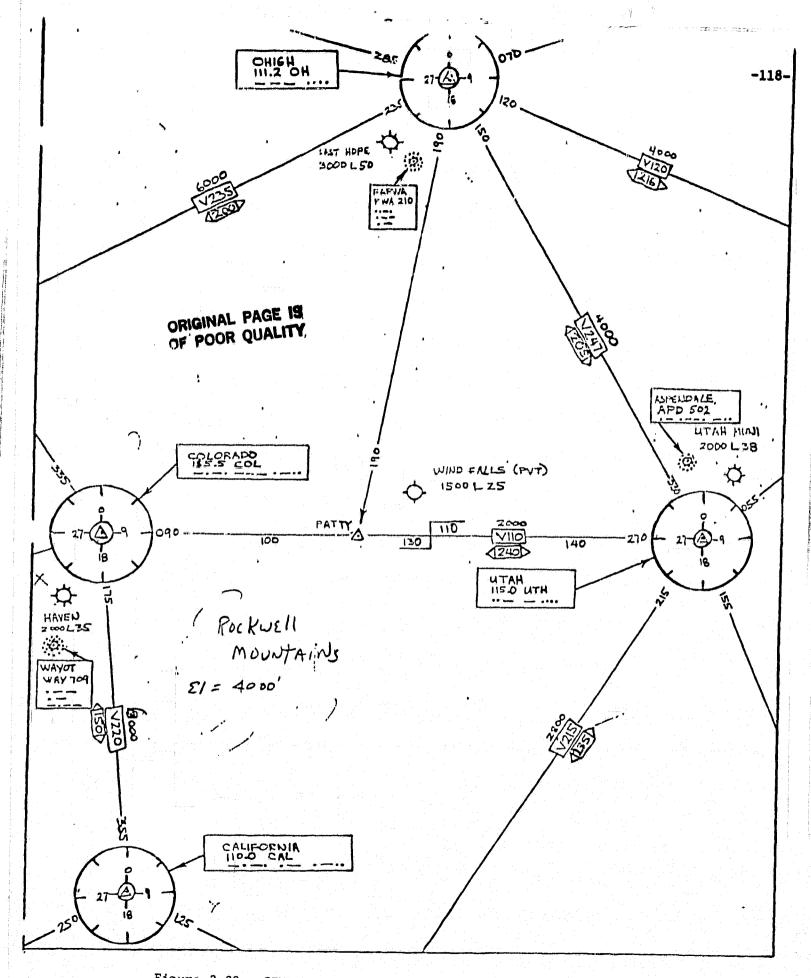


Figure 3-39: SIMULATED LOW ALTITUDE CHART

| R | EQUEST OF F | TC INFO | | | | | | | |
|---------------------------------|-------------------------|-----------|------------|------------------------------|-----------------------------------------------------------------------------------------------------------------|-----------|--|--|--|
| 隵 | Ceiling | Visi | bility | Cloud Tops | Winds | Aloft | | | |
| | PIREFS | SIC | SI1ETS | AIRHETS | | | | | |
| | Ground Speed | | AID atus |] Freezing Level | a a construction and a construction of the fraction of the second second second second second second second sec | | | | |
| COMMUNICATION WITH ATC Filot is | | | | | | | | | |
| Pil | ot requests | | | declaring . | an emergency | ul and | | | |
| | heading ch | ange | deg | changing h | g eading | deg | | | |
| | altitude c | hange | ft | changing a | Øltitude | ft | | | |
| | Confirm ne after you | | and altitu | de Headin Altitu | - | deg ſt | | | |
| | Pilot woul | d like to | | of a problem altitude cha | mand | | | | |

Figure 3-40: ATC COMMUNICATION DISPLAY

The scenario used in this experiment is a modification of scenario 5 described in Chapter 2. Essentially, it was decided to give the pilot a problem which would not likely be solved within the time available. A broken baffle was selected to ultimately force the pilot to consider his decisions about selection of a destination airport.

The simulated full mission scenario located the pilot such that he was heading into the mountains with an unresolved problem, i.e. unable to get full power from the engine and required by ATC to maintain 5000 feet in IMC conditions. Figures 3-41 through 3-45 depict the typical development of the situation with ATC communication to enhance position awareness. Table 3-13 investigates some of the inquiries made by a pilot attempting to resolve the problem.

Results

Table 3-14 describes the various responses made by pilots using this "full mission scenario". The intent of this study was to force a choice dilemma situation, i.e. the pilot could elect to: (a) struggle on over the mountains with a sick engine hoping no other problems develop, (b) declare an emergency and seek to land the aircraft immediately in minimum VFR conditions at a nearby private airport, or (c) to return to the airport of origin where instrument aids would be available. It was also of interest to ascertain how long the pilot would attempt to diagnose his problem before accepting the fact that he would be unable to get full power from the engine.

DIAGNOSTIC SCENARIO TEST

We are now going to present to you some Critical In-Flight Events requiring your diagnosis of the problem.

Assume that you are flying a fuelinjected Cherokee Arrow (N123B) with the following performance specifications:

Cruise Speed = 135 KTAS (65% pwr. @ 7000 ft.)

Fuel Flow (65% pwr.) = $1\emptyset$ GPH

Usable Fuel Capacity = 48 gallons

Endurance = 4.8 hours (no reserve)

Range = 648 nautical miles (no wind, no reserve)

Press CONTINUE when finished reading.

CONTINUE

3

Figure 3-41: SCENARIO INITIALIZATION

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Consult attached simplified low altitude chart.

You are on an IFR flight from Utah Municipal Airport to Haven County Airport. You depart on V-110 at 6000ft in your Cherokee Arrow (N123B) which is equipped with a 3-axis autopilot. There is a NOTAM out which reports that Colorado VOR is out of service during the period you plan to navigate. <u>Navigate using Ohigh</u> and California VORs. You have been enroute 60 minutes from Utah Municipal Airport. You are on the gauges but the ride is smooth. Weather briefing indicated that winds at 6000 were expected to be light and variable.

You have one passenger aboard.

Weather at:

Haven County Airport= 2000 & 5 Ohigh= 1000 & 3 Wind Falls= 1000 & 3 by a C-172 (10 minutes ago)

Cleve Center calls and reports radar contact is lost. Please report present position.



When ready, press the CONTINUE button to go to the VOR display to establish position.

POSITION REPORT

Please report your position by pressing the TO or FR buttons for the VOR of your choice. (Choose at least two VORs).

Type in your position via the keyboard at the given arrow, then press the NEXT key to enter it.

| California VOR | 1701 | 4Ø | ok | | |
|--------------------|------|----|----|-----|----|
| Utah MunAir VOR | Tol | | | 260 | ok |
| Ohigh VOR | 170) | | | | |

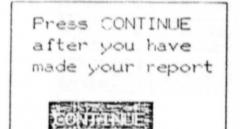


Figure 3-43: POSITION REPORTING

Last Clearance

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ATC Response:

N123B, thanks for the position report. Here is your new clearance: proceed direct California VOR direct Haven County Airport at 6000.

There will be opposite traffic at 5000...maintain 6000.

Please confirm your new heading and altitude after your turn.

| - Velt H | Callan - |
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| | ENTRY THE AVERAGE AND |
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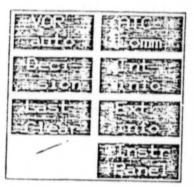
Time: 7:31 Scenario: 05

SCENARIO CHANGE

While practicing hand flying with your autopilot disengaged, you notice that increased nose-up trim is required to maintain a constant indicated altitude and that your IAS has decreased 20kts. from normal cruise.

Your passenger notes this problem, and suggests that you turn back to Utah Municipal.

Determine the nature of the problem, and your destination decision.



Time: Scenario: Ø5

Figure 3-45: INTRODUCTION OF THE CIFE INTO THE SCENARIO

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| DATANDI ##cif(##sce | e3 | | NAM Scenario #05 DAT |
|----------------------------|------|-----------|-------------------------|
| TIME | TIME | DISPLAY | ITEM |
| Ŭ, | 10 | Int info | |
| 10 | 3 | Int info | cargo cond |
| 13 | 4 | Int info | door cond |
| 17 | 4. | Int info | panel temp |
| 21 | 4 | Int info | noise & vibration |
| 25 | 3 | Int info | fluid leaks |
| 28 | 3 | Int info | housekeeping |
| 31 | 2 | Int info | smoke |
| 33 | 6 | Int info | cabin temp |
| 39 | 3 | Ext info | |
| 42 | 4 | Ext info | cowling |
| 46 | 2 | Ext info | windscreen |
| 48 | 23 | Ext info | Wing |
| 51 | 2 | Ext info | flap |
| 53 | 2 | Ext info | aileron |
| 55 | 18 | Ext info | stabilizer |
| 73 | 7 | instr pan | • |
| 80 | 2 | instr pan | pitot heat |
| 82 | 12 | instr pan | pitot heat on |
| 74 | 17 | instr pan | alt static |
| 111 | 18 | instr pan | suction |
| 129 | 2 | instr pan | pitot heat |
| 131 | 15 | instr pan | pitot heat on |
| 146 | 13 | instr pan | TACH |
| 159 | 14 | instr pan | MP |
| 173 | 18 | VOR-Auto | |
| 191 | 55 | Scenario | |
| 246 | | GIVE ANSR | |

NAME:Estudent082 DATE: 07/30/82

> CURRENT VALUE

Table 3-13: Illustrative Pilot Inquiries in the Combined Scenario

Early trials indicated a complete preoccupation with diagnosis and little position awareness. Attempts were made to increase position awareness through position reports to ATC and a passenger insisting they return to the origination airport. Despite these efforts, most pilots indicated little concern for altitude clearance violations and spent most of the time wrestling with the engine problem. Most opted to continue over the mountains without appraising ATC of the problem, though some pilots did declare emergencies or return to the airport of origin. Despite the fact that some pilots had experienced this engine problem earlier in diagnostic tests, there was little evidence that they recognized this problem or abandoned the diagnostic inquiries earlier than pilots viewing this problem for the first time. Table 3-14 shows that there is a considerable variation in diagnosis time, number of inquiries, and return to old track inquiries (DIFT). Only one pilot declared an emergency and none elected to put the aircraft down at the nearby private strip. Table 3-15 reflects performance differences of subjects who took scenario 5 as a pure diagnosis problem as compared to those who took it in conjunction with the destination diversion decision (denoted as P-G on PLATO® GAT). Again we see a wide variety of responses from both groups. The P-G group took more time on the first track and made more inquiries in general. This probably reflects the stress of the navigation decision.

The general conclusions from this experiment were:

 (a) using CAT it is possible to provide realistic displays for navigation purposes;

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| | | | - | | | | | | | | | - | - | | | i |
|---|-----------|----------------------|--------------------|-----------------|----------|-------------------------|------------------------|-------------------|-----------|-----------------|-----------------|------|-----------|--------------------|---------------|-------------|
| | MAX | | | m | 6 | 1 | 1 | 650 | 69 | 25 | 6 | 17 | 3 | 6 | 2 | |
| _ | 84 | Haven | >90 | 0 | F | N | Gyro Failure | 324 | 10 | 'n | 4 | T | 2 | 2.5 | 2 | - |
| | 83 | Haven | 30-60 | 0 | 0 | N | Power | 404 | 37 | 19 | 6 | 10 | 1.95 | 4.111 | 0 | |
| | 82 | Haven | 30-60 | 0 | 0 | N | Ďon ' t Know | 246 | 23 | 12 | 7 | 5 | 1.92 | 3.286 | 0 | |
| | 8.1 | Calif. | 06-09 | ō | 0 | N | Don't Know | 204 | 42 | 25 | 00 . | 17 | 1.68 | 2.47 | 0 | · · · |
| | 80 | Calif. | 06-09 | (Alt Hold) 3 | 0 | N | Don't Know | 647 | 63 | .24 | 7 | 17 | 2.625 | 6 | 7 | |
| | 7.9 | Continue | 06< | 0 | 0 | N | Elevator Broken | 0 | T | T | Ţ | 0 | T | 1 | 0 | |
| | 78 | Utah Muni | 06-09 | 0 | 6 | Τ | Engine Temp High | 650 | 69 | 25 | 8 | 17 | 2.76 | 8.625 | 0 | |
| | 77 | Haven | 06< | 0 | 0 | N | Don't Know | 436 | 30 | 17 | ۷ | 10 | 1.76 | 4.286 | و | |
| | 76 | Haven | 0-30 | 0 | 0 | N | Power Lost | 237 | 27 | 6 | 9 | 3 | 3 | 4.5 | 0 | |
| | Subject # | Airport Selection | Time to Airport | Alt. Changes | ATC | Declared Emergency ? | Diagnosis | Diagnosis Time | Tot. Inq. | Total Tracks | Uniq. Tracks | DIFT | Inq/Track | Inq/Uniq. Track | Yoke Resp. | |

Table 3-14 Summary of Performance in Combined Scenario

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ORIGINAL PAGE IS OF POOR QUALITY Tracks in Scenario 5 Alone and in Plato-Gat Table 3-15

| Subject # | Scen 5 or Plato Gat | First Track | <pre># Inq. on 1st Track</pre> | Time on lst Track | 2nd Track |
|------------------|---------------------------|-------------------------------|--------------------------------|----------------------|-----------------------------|
| 56 IFR | 5 | Struct. Fail. | 2 | 27 | Other (gear) |
| 57 IFR | 5 | Other (Suction) | 1 | 13 | Ice |
| 58 IFR | 5 | Int. Enq. (MP) | 1 | 16 | Ice |
| 59 IFR | 5 | Ice | 1 | 16 | Flight Status Struct. |
| 60 VFR | 5 | Other (Cargo Cond) | 1 | 27 | Fail. Other |
| 61 VFR | 5 | Ice Other | 4 | 31 | (ceiling) |
| 62 VFR | 5 | (Cargo Cand) Other | 2 | 41 | Ice Struct. |
| 63 VFR | 5 | (Cargo) | 2 | 46 | Fail. Prop |
| 64 VFR | 5 | Ice | 2 | 26 | (TACH) |
| 65 VFR | 5 | Int. Eng. | 3 | 19 | Fuel Other |
| 66 VFR | 5 | Ice Other | 3 | 26 | (groundspeed) Prop |
| 67 VFR | 5 | (Cargo) Other | | . 11 | (N&V) Prop |
| 68 VFR 69 IFR | 5 5 | (Suction) Other (Cargo) | 1 | 7 | (TACH) Ice |
| 70 IFR | 5 | Other (Gear) | 2 | 59 | Fuel (Throt.) |
| 76 VFR | PG | Int. Eng. | 5 | 49 | Ice |
| 77 VFR | PG | Struct. Fail. | 3 | 44 | Ice Other |
| 78 IFR | PG | Flight Status | 2 | 207 | (Flaps) |
| 79 IFR | PG | - | 0 | 0 | Other |
| 80 IFR | PG | Ice | 4 | 31 | (Suction) |
| 81 VFR | PG | Fuel Other | 4 | 26 | Ice Prop |
| 82 VFR | PG | (Cargo) | 3 | 27 | (N&V) |
| 93 IFR | PG | Int. Eng. ATC | 3 | 24 | Fuel Flight |
| S4 IFR | PG | (DG HEAD) | 2 | 211 | Status |

(b) most pilots are reluctant to accept the fact that no diagnosis or solution is available in flight. This may result from the fact that all other scenarios presented had definitive solutions;

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- (c) diagnosis occupied the pilot's time much more than positional awareness;
- (d) considerable variation in pilot response to this "full mission scenario" was apparent;
- (e) no clear relationship between pilot information seeking and decision making and pilot experience and training was noted (probably due to the small sample sizes).

In effect, this experiment attenpted to produce a GAT simulator experience using a computer terminal. Upon subject debriefing it was apparent the pilots were overloaded with information and reacted to the problem by treating the engine symptoms and navigation problems independently as opposed to an integration of diagnosis and diversion information. In their defense the subjects argued that in a real aircraft, positional (navigational) and engine information would be simultaneously available. They also argued that a one time presentation of this kind of scenario gave them little chance to adapt, i.e. they had little chance to do VOR tracking prior to the CIFE. Unfortunately, real CIFE's give pilots little chance to adapt.

There was insufficient time or funding available in the research program to pursue this "full mission scenario" using CAT beyond these exploratory experiments. It may be that trying to produce a type of GAT-1 simulator using computer disclays is not an efficient or realistic approach.

Despite the question of whether this approach has value in understanding pilot logic structures, there appears to be considerable potential for pilot training. ť.

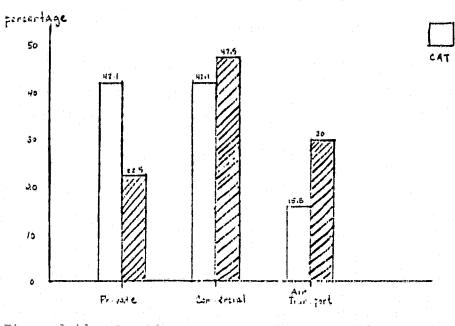
I. Comparison of Computer Aided Testing With Paper and Pancil

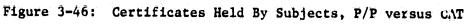
The four basic diagnostic scenarios and the destination diversion scenario used in computer aided testing (CAT) were based on the same underlying problems as those used in earlier paper and pencil (P/P) experiments. The principal differences involved manner of presentation and timing. CAT was nearly experimenter free. Subjects received instructions and sought information from an interactive computer graphics display. The computer recorded the sequence of information exchanges and the time between inquiries. The P/P tests on the other hand required a verbal exchange between experimenter and subject to transfer information. Although the experimenter did record the order of inquiries, there was no provision for obtaining a complete time history.

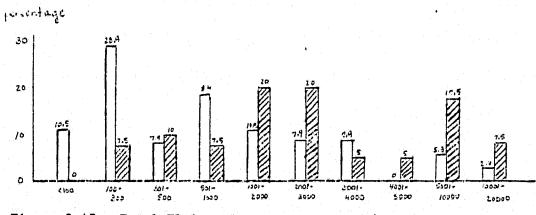
Subject Characteristics

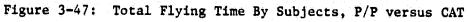
The sample sizes for CAT and P/P testing were nearly identical. CAT used 42 pilot subjects while P/P used 40. A quick comparison of the relative flying experience of the two groups can be seen in Figure 3-46, 3-47, and 3-48. Their relative performance on the knowledge test is shown in Figure 3-49. Both groups had identical questions on the knowledge test instrument.

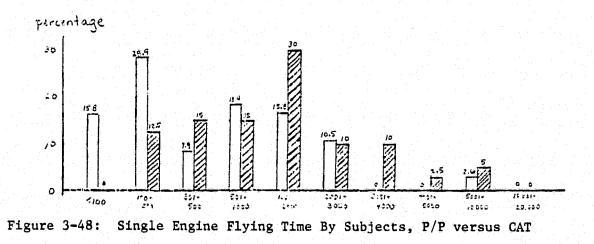
The general impression is that the CAT test group is less experienced. The CAT group has more private pilots, lower total flying time and lower single engine time than the P/P test group. The entire group of P/P subjects hold the instrument rating while sixteen pilots in the CAT group are not instrument rated. The P/P group scored slightly higher in the knowledge test. These characteristics are summarized in Table 3-16.



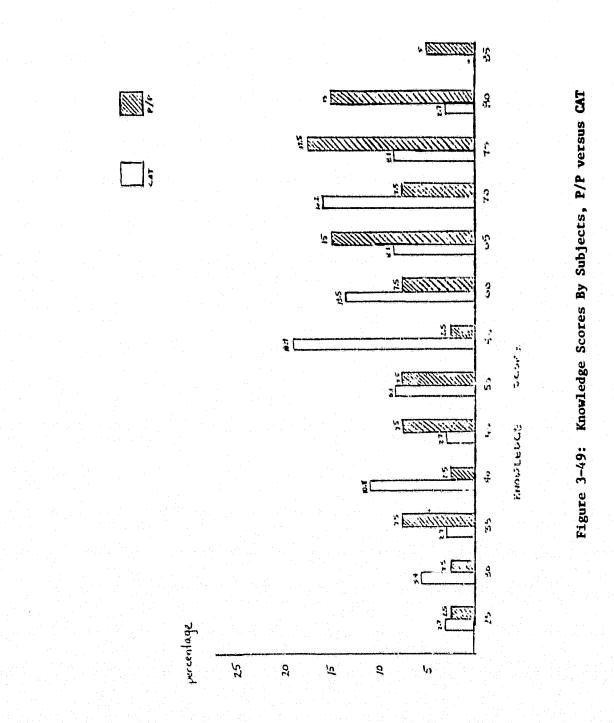








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| ÷ | | RATINGS (Z) | | | | AVG | KNOWLEDGE |
|-----|----------|-------------|------|-----|-----------|------|-----------|
| | SUBJECTS | PVT | COMM | ATP | AVG TT | SET | TEST (%) |
| CAT | 42 | 42 | 42 | 16 | 1654 | 933 | 57 |
| P/P | 40 | 22 | 48 | 30 | 3821 | 1911 | 62 |

Table 3-16. CAT and P/P Group Characteristics

Diagnostic Performance

Correctness scores for each of the four basic scenarios were based on five points for a perfect answer. However, the grading mechanism did change slightly between the P/P and CAT experiments. For that reason the relative differences among scenarios may be more significant than raw score differences between test groups. The average scores are depicted in Table 3-17. It should be noted that all P/P subjects took every scenario while not all of the CAT subjects did. Consequently, adding average scenario scores for CAT will not yield the total average shown.

| | AVG. PERCENT | Avera | age Scenarios | Correctness | Scores | |
|-----|---------------|------------|---------------|-------------|------------|--|
| | TOTAL CORRECT | SCENARIO 1 | SCENARIO 2 | SCENARIO 3 | SCENARIO 4 | |
| CAT | 47 | 3.3 | 1.5 | 3.3 | 3.0 | |
| P/P | 65 | 2.7 | 2.9 | 3.4 | 4.0 | |

Table 3-17. Correctness Score Comparison

The higher average score for the P/P group is indicative of the generally higher level of flying experience of that group compared to the CAT group. An interesting finding occurs in the relative rank of scenario 2 scores. Scenario 2 involves a vacuum pump failure and subsequent loss of gyro instruments. The P/P group scored relatively higher on that scenario than did their CAT counterparts. Since the CAT group contains a number of non-instrument rated pilots, this result may suggest that pilots who must routinely depend on their gyros are more cognizant of unusual indications. Table 3-18 compares general search patterns between the two experiments. The interesting thing here is that, with the exception of scenario 1, the P/P group made far fewer inquiries per scenario than did the CAT group. As one might expect, the CAT group also explored a higher number of tracks. This may suggest that the computer aided testing is more "user friendly" and invites a broader search for alternatives than does the verbal exchange in a paper and pencil format. One might also speculate that computer users are less inhibited about asking for what may be frivolous information than are their verbal counterparts who risk sounding foolish to the human experimenter who must respond to their questions.

Regression analyses performed on CAT scores reveal that knowledge subscores and source of flight training are predictors for correctness scores on some scenarios. These results are consistent with regression analyses performed on the P/P data. The only striking difference between the groups is that correctness scores by scenario seem to be related for selected pairs among CAT subjects while in no case is a good score on scenario i a predictor of score on scenario j in the case of the P/P subjects.

When subjects are split into high and low groups for t-test comparisons, results of CAT and P/P tests are gimilar. For example, in both cases subjects with high knowledge test scores tended to achieve high correctness scores and they were also the pilots with the most single engine flying time. Since most of these tests are based on the same biographical data and knowledge test scores,

| | CAT | P/P | |
|---------|-------|------|--|
| TOTINQ1 | 6.12 | 6.38 | |
| TOTRAK1 | 3.33 | 1.80 | |
| UNTRAK1 | 2.57 | 1.50 | |
| TOTINQ2 | 15.21 | 8.58 | |
| TOTRAK2 | 6.76 | 3.75 | |
| UNTRAK2 | 3.57 | 2.75 | |
| TOTINQ3 | 15.10 | 9.58 | |
| TOTRAK3 | 7.14 | 3.85 | |
| UNTRAK3 | 4.00 | 3.15 | |
| TOTINQ4 | 13.90 | 9.12 | |
| TOTRAK4 | 5.76 | 4.10 | |
| UNTRAK4 | 3.34 | 3.35 | |

Table 3-18: DIAGNOSTIC INFORMATION SEARCH MEAN VALUES

the lack of glaring differences offers evidence that CAT and P/P testing are similar in performance measures expected from any given class of subjects.

Destination Diversion Performance

The difference in the mechanics of information exchange between CAT and P/P testing is again reflected in the inquiry statistics for the destination diversion scenario. The average number of inquiries is 22 for CAT versus 17.2 for P/P subjects. The information requests per airport shows a similar trend with averages of 3.82 versus 2.60 for CAT and P/P, respectively. The interactive terminal either invites a wider search or makes the information exchange more efficient thus permitting more inquiries in a fixed test period.

The general pattern of airport consideration between the groups is similar in that the frequency of consideration for a given airport decreases for airports further away. Both CAT and P/P subjects showed heavy interest in airport 5. However, the final choice of a destination diversion airport is completely different. The highest ranking airport among the P/P subjects was airport 14 while the CAT subjects preferred airport 3. Both choices have similar facilities (i.e. VOR approach and 4 tower) and both are in the same general direction. They differ in that 3 is in hilly terrain while 14 is in the mountains. The final choice may have been more influenced by the form of the map than by any real difference in preference between the groups.

The relative frequencies of type of information requested are quite similar for both groups. Ceiling, visibility and approach aids are the most often requested information items. The medium of presentation does not generally appear to affect preferences for type of data in the destination diversion problem. There is one unexplained anomaly, however. Bearing and distance information represents 14.4 percent of the total information requests from CAT subjects and only eight percent from P/P subjects. It is not clear why this data should be a sizeable portion for either group since it can be at least roughly estimated simply by glancing at the enroute chart without wasting time on an information query. Chapter IV: Learning In Computer Aided Testing of CIFE Diagnosis

Task 7 of the research was directed to subjects learning in terms of performance changes both within test sessions and across test sessions. The significance of this task can be better appreciated when one considers the long range potential of incorporating CAT in pilot training. Self-paced training would enable pilots to proceed through scenarios of increasing difficulty until some level of competency was accomplished.

Because of the novelty effect in using touch CRT's there was a need to see if pilot performance in terms of correct diagnosis, efficiency in arriving at diagnosis, and information seeking style would change within a test session and across test sessions. Insufficient time prevented across test learning experiments.

Within Session Learning

In order to test for order effects and not confound scenario difficulty with order, two order sequences were used. One group of subjects received scenarios 1, 2, 3, and 4, in that order. A second group were tested in the order 2, 3, 4, 1. Measures of performance included:

- (a) the total number of inquiries
- (b) the total number of tracks
- (c) correctness score
- (d) the time interval between inquiries (mean and σ^2)
- (e) the number of unique tracks

(f) the total time to complete the diagnosis

In order to evaluate aggregate learning on performance scores it was first necessary to normalize subject data. Data for each subject was normalized by taking the average score across the scenarios

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for each subject and then dividing individual performance measure by the average to get percentages. For example, for a given subject, correctness score on a given scenario would be some percentage of his average correctness score on all scenarios. This technique enables subject data to be added without giving heavy weighting to subjects with large numerical scores.

Table 4-1 shows a two way ANOVA - scenario vs. order for three scenarios (2, 3, and 4) and two orders (1,2,3,4, vs. 2,3,4,1). Note that for most performance measures the order of testing had no effect. As might be expected, scenarios were somewhat different in terms of correctness, number of inquiries, number of tracks, and number of unique tracks. It was generally found that scenario #2, the vacuum problem was the most difficult scenario for the subjects.

When time data is examined, means of time between inquiries (delta time) show a pronounced trial effect, i.e. the first trial (for both orders) is higher than all the remaining trials. Figure 4-1 separates IFR and VFR pilots for each order and depicts changes in mean delta times over the trials. The reason IFR and VFR pilots were separated was because of possible confounding effects. VFR pilots are over-represented in the order 2-3-4-1. With the IFR-VFR break out it is still clear that learning takes place across trials within each order in terms of time between inquiries. Beyond that finding, little learning is evident across trials within an order. This is an interesting and reasonable finding. Reduction in delta times comes with familiarity in using PLATO[®]. Diagnostic performance should be dictated by scenario difficulty and not by order of testing. From Figure 4-1 it appears that learning in terms of delta time would continue beyond the four scenarios tested.

e i.#.

| Depend. Var. | Source | F Value | Pr > F | ·Signif.? |
|----------------------------------------|---------------|------------|----------------|-----------|
| Correctness | Order | .01 | 9129 | No |
| | Scen | 2.20 | .1175 | No |
| # of | Order | .45 | .5059 | No |
| Inquiries | Scen | 2.08 | .1313 | No |
| # of | Order | .26 | .6093 | No |
| Tracks | Scen | 3.08 | .0513 | Yes |
| # of | Order | .26 | .6090 | No |
| Uniq. Tracks | Scen | 5.44 | .0061 | Yes |
| DIFT (# Tracks - # Uniq. Tracks) | Order Scen | .07 .47 | .7896 .6293 | No No |

Table 4-1 Order & Scenario Effects on Performance

Mean of the delta times

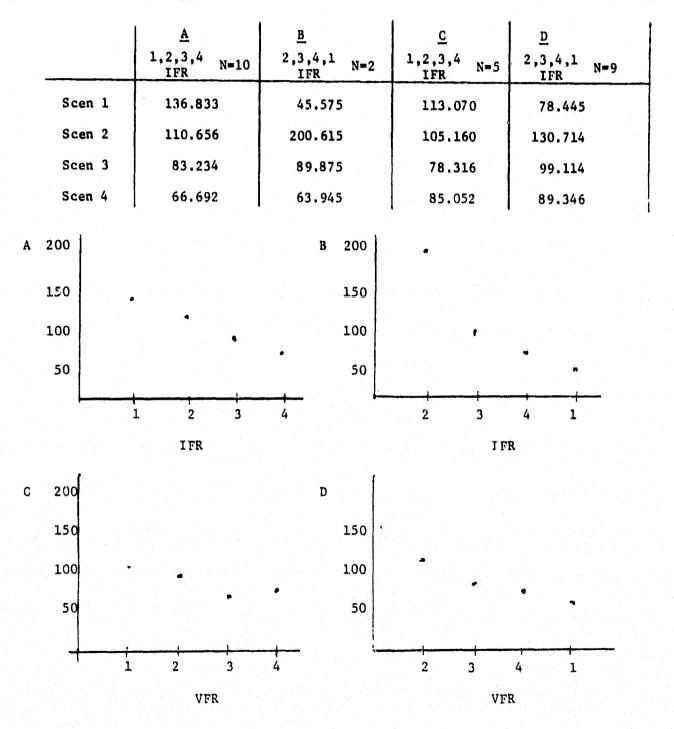


Figure 4-1 Changes in mean delta times by type of rating and order of scenario testing

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To test for a confounding of rating with order, contingency tests were executed in a two by two design (two orders and two ratings - VFR/IFR) for the various performance measures. In all cases the hypotheses of independence were rejected at the ten percent significance level.

Chapter V: Computer Aided Programming

A wide variety of information search patterns were exhibited by subjects during CIFE diagnoses, some of which were more logical and efficient than others. In an attempt to streamline search patterns, preliminary investigation of the use of a diagnostic computer aid was carried out. A prototype Decision Support System (DSS) to aid a pilot in his decision making during an emergency situation was designed. The suggested model, called the Airplane Condition Evaluation (ACE), was patterned after MYCIN, a diagnostic computer aid used by physicians in their diagnosis of blood diseases.

ACE uses an interactive program similar to MYCIN which utilizes production rules and allows the user to input the current facts of the situation. Each of these production rules contains a small amount of information for ACE to draw upon. ACE is able to "reason" with current facts and to "deduce" a conclusion by joining together a series of these production rules. The production rules are modular and may be joined together in any way that the current facts support, thus allowing ACE to generate a diagnosis with any combination of aircraft symptoms. This system is described in complete detail in the NS thesis by Jeffrey A. Lee, A Decision Support System For In-Flight Emergencies: ACE (5). Mr. Lee is a Graduate Research Associate on the NAG 2-112 research project. A copy of that thesis has been submitted along with this report and is available at NASA Ames.

ACE is a "suggestion model" type of DSS which plays the role of a consultant with the expertise of an experienced aircraft pilot. It has the capability of accepting many variations of the problem it is geared for. Since it is computer driven, it has the advantage

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of speed which makes it particularly well suited for a real-time environment. Such a system could be physically located in the cockpit of more sophisticated aircraft or located on the ground for use by smaller aircraft. In the ground mode some means of data exchange, perhaps as simple as voice radio communication, would be required. If the system was located at a Flight Service Station, for example, any flight service specialist could serve as a surrogate for the expert consultants currently available to air carrier and military. ACE would guide the questioning which the FSS specialist could relay to the pilot.

In its current state of development ACE contains a total of fifty-one production rules. These rules were distilled from a frequency count of information requested by PLATO® subjects who correctly diagnosed scenario problems and augmented by discussion with local "expert" pilots. The experts were used to establish the reason why information items requested with high frequency led to a correct diagnosis. The experts were able to suggest what could be inferred from a group of items of information and also what additional information would be needed to further support the line of reasoning. This analysis gave way to the formation of production rules and thereby established the expert knowledge and reasoning necessary to draw the correct diagnosis for a CIFE.

ACE has so far been constrained to the four scenarios developed in the earlier paper and pencil studies and later extended to the PLATO® tests. ACE does not yet have the ability to work in the real world but only in conjunction with this set of four scenarios and the corresponding set of information for each as presented by the PLATO® data-gathering computer system developed for this research. Within those constraints, however, the 51 production rules contain enough expertise to correctly diagnose each of the four scenarios. -146-

The following sample session with ACE illustrates a typical set of computer queries and subject responses. In testing ACE, responses were made consistent with data the subject obtained by querying the PLATO® representation of current aircraft status. Thus, two computer systems were used, a DEC 20 to house ACE and the CDC host computer which houses PLATO®. The accompanying logic trace shows the trees used in chaining to the recommended action or diagnosis.

The following is a sample session with ACE.

*** Welcome to ACE... the Airplane Condition Evaluator

ACE will try to diagnose a Critical In-Flight Event by asking you a series of True/False questions. Flease answer each of the questions with a "T" for a true statement or an "F" for a false statement. You may also ask ACE why it is asking you a particular question by entering "WHY", ACE will respond with the production rule that it is currently considering to create a probable diagnosis of the CIFE.

Flease enter "T" to start the questioning. T

SUCTION LOW? (T/F) WHY

*** HYPOTHESIS<1> IF: VACUUM PROBLEM OBV: SUCTION LOW ACT: VACUUM FAILURE

SUCTION LOW? (T/F)

...

1

AS NOT CONSTANT? (T/F) Ť. ALT NOT CONSTAN' (T/F) T (VSI NEGATIVE OR VSI POSITIVE)? (T/F) Ť AH LEVEL? (T/F) DG CONSTANT? (T/F) T T/B SHOWING BANK? (T/F) T AIRFLANE CRUISING? (T/F) Т AS INCREASING? (T/F) Υ ALT DECREASING? (T/F) T THE COMPUTED DIAGNOSIS IS THE FOLLOWING: -> VACUUM FAILURE DESIRE TRACE OF LOGIC? (T/F) T DESIRE TERSE TRACE? (T/F) T TREE<1> IF: STOP OBV: AIRPLANE CRUISING DEV: AS INCREASING OBV: ALT DECREASING OBV: AH LEVEL ACT: FOSSIBLE VACUUM FROBLEM TREE<13> IF: FOSSIBLE VACUUM PROBLEM DEV: AS NOT CONSTANT DEV: ALT NOT CONSTANT

OBV: (VSI NEGATIVE OR VSJ POSITIVE) OBV: AH LEVEL OBV: DG CONSTANT OBV: T/B SHOWING BANK ACT: VACUUM PROBLEM

HYPOTHESIS<1> IF: VACUUM PROBLEM OBV: SUCTION LOW ACT: VACUUM FAILURE

ACE's facility for showing the current rule that it is considering and its ability to show a trace of the rules it used to compute a diagnosis make the system more user friendly. A user sees not only where a current request for information came from but also may be reminded of possibilities that might not have occurred to him.

ACE is a prototype of a computer aid that may be useful in actual in-flight emergencies. It demonstrates that there is potential in the idea that a diagnostic computer aid can be built with production rules and be flexible enough to handle different scenarios and their individual sets of information.

Chapter VI: Summary and Conclusions

The tasks set forth in the proposal on the "Use of Computer-Aided Testing in the Investigation of Pilot Response to Critical In-Flight Events" have been accomplished. This project continued the early concern for the dynamics of CIFE's explored in the first report for NAS 2-10047 (6).

The products of the current research are 1) the First Symposium on Aviation Psychology held at Ohio State in 1981 (4); 2) a set of programs including new scenarios for computer aided testing (CAT) of subjects in CIFE's; 3) a dynamic system for CAT permitting combined diagnostic and destination diversion decisions (PLATO®-GAT); 4) a prototype Decision Support System to streamline search patterns (ACE); 5) a data base of responses from 40 pilots in the Columbus area (80 total when added to earlier paper and pencil subjects). By-products of this research so far have included one M.S. Thesis, two journal publications and a conference proceedings.

A. First Symposium on Aviation Psychology

The First Symposium on Aviation Psychology was held at Ohio State on April 21 and 22, 1981. The symposium was supported by this NASA grant. The objective of the symposium was to "critically examine the impact of high technology on the role, responsibility, authority and performance of human operators in modern aircraft and air traffic control systems." (4-i)

The attendee list for the symposium contained 210 persons from across the nation, many with outstanding credentials in aviation, engineering, or aviation psychology. Over 40 papers were presented at the symposium and published in the proceedings. In addition,

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selected ones of these papers were published in the <u>Journal of</u> <u>Aviation Space and Environmental Medicine</u>. As a result of these efforts a second symposium has been scheduled for April 1983.

B. Computer Aid Testing Demonstration

The research team was successful in converting paper and pencil scenarios for diagnosis and destination diversion decisions into a computer interactive touch CRT system. This system provides the following advantages over the earlier paper and pencil studies.

- 1) The system is experimenter free, i.e. detailed instructions are given via the graphics display permitting subjects to run the system without the physical presence of an experimenter.
- 2) Subjects are enthusiastic about the presentation format. Their motivation remains high for test periods of up to 90 minutes long.
- 3) Subject response requires only a simple touch of the CRT screen. Keyboard entries are not required after initial sign-on.
- 4) Menu response formats reduce ambiguity and facilitate data analysis.
- 5) A complete catalog of pilot responses is available in the data display module.
- 6) A complete time history of pilot inquiries and responses is recorded.
- 7) National testing capability is present without moving experimenters or equipment. The many CDC Learning Centers throughout the U.S. as well as privately held PLATO[®] terminals can be used to call up and run the programs on site. Data files are maintained centrally.

From comments made by pilots who participated and from observations by the research team, it appears that computer aided testing (CAT) offers a realistic method to study pilot response to critical inflight events. In addition, CAT has the potential to become a valuable training aid.

C. The PLATOD-GAT System

The term "PLATO?-GAT was coined to describe the dynamic navigation and control system developed for computer aided testing. The task to be addressed was one of combined destination diversion and diagnosis decisions. The PLATOR-GAT system demonstrates the following capabilities:

- 1) The system provides an automatic tracking feature which permits the experimenter to place the subject aircraft anywhere in the simulated test area, then track the movement of the aircraft over time.
- The pilot can determine his position in space by crosschecking on dual VOR indicators. The indicators are dynamic and show course deviations over time.
- 3) The pilot can alter his simulated heading and altitude by appropriate touch-panel entry on the auto-pilot display.
- 4) ATC position reports and clearances are part of the scenario.
- 5) The system combines in the touch CRT both the destination diversion decision and the problem of diagnosis.

The intent of PLATO GAT was to introduce a more realistic framework for tests which would begin to approach the fidelity of earlier GAT simulations. The dynamics associated with the decision making portion of a GAT-type simulation experiment were preserved without the accompanying stick and rudder manipulation. This was accomplished on a simple set of CRT displays.

D. Computer Aided Prompting

A prototype decision support system to aid pilots in decision making during emergency situations was developed. The model called Airplane Condition Evaluation (ACE) was patterned after MYSIN, a diagnostic computer aid used by physicians in the diagnosis of blood diseases.

ACE has the following characteristics:

1) It is an interactive program which permits user input on the

current facts of a situation.

2) ACE "reasons" with current inclusion by joining together a series of production rules.

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- 3) Although designed around the four basic CAT scenarios, ACE has the capability of accepting many variations of the problem it is geared for.
- 4) ACE is user friendly in that it shows the current rule being considered as well as a trace of the rules it used to complete a diagnosis.
- E. Results from CAT

A series of correlation studies, stepwise regression analyses and t-tests were performed on the combination of pilot background variables, knowledge survey results, diagnostic scenario performance and decision making measures. Among the observations made from these analyses are the following:

- 1) Of the four basic diagnostic scenarios tested, pilots have the most difficulty in identifying the symptoms of a vacuum pump failure.
- 2) Pilots who score well in the knowledge test are good at diagnosing vacuum pump and static system failures.
- Knowledge is inversely related to diagnostic inquiries, i.e. knowledgeable pilots reach conclusions (right or wrong) more rapidly than others.
- 4) Less experienced pilots tend to use a larger number of diagnostic tracks than do more experienced pilots.
- 5) Civilian trained pilots score higher on the vacuum pump failure diagnosis than do military trained pilots.
- 6) IFR rated airmen receive higher knowledge scores and higher diagnostic correctness scores than do VFR rated airmen.
- 7) High correctness scores are positively related to high mean time between inquiries, i.e. pilots who take time to absorb and manipulate current information before seeking new data are more apt to be successful than those who rapidly collect large amounts of disjoint information.
- The data most often requested in diversion decisions are ceiling and visibility. Terrain receives a low number of inquiries.

- 9) Pilots often neglect winds aloft information in selecting an alternate airport.
- 10) There is no discernible statistical difference in the time or inquiry performance of professional and non-professional pilots in selection of alternate airports.

In addition to the more formal statistical comparisons, special graphical devices were used to analyze pilot performance. These devices were used to analyze pilot performance. These devices were the pilot information plots (PIP), schema diagrams, and destination graphs (DIG). Among the observations made from these analyses are the following:

- 1) Pilots follow a wide variety of different search patterns during diagnosis.
- 2) Pilots tend to exhibit similar search strategies across all scenarios, i.e. a pilot with a very systematic search pattern on any one scenario will behave similation on all others.
- 3) Based on PIP and Schema Analyses, efficient information searching can be hypothesized. It appears that the ideal pilot should do the following:
 - i) Confirm symptoms.
 - ii) Establish engine health status.
 - iii) Establish an hypothesis concerning cause.
 - iv) Determine hypothesis plausibility with a minimum number of inquiries on an appropriate track.
 - v) Given logical cause of the symptoms, test alternative hypotheses by additional information inquiries.
- 4) Schema diagrams are scenario dependent.
 - In the oil leak scenario, the only distinguishing feature of correct versus incorrect groups is that the correct group focused on internal housekeeping questions while the incorrect did not.
 - ii) In the vacuum pump failure scenario, the only common element between the correct and incorrect groups is a strong concern for icing indications. Only the correct group bothered to check gyro instruments.

- 111) In the broken magneto scenario, the correct group checked the ignition system while the incorrect one did not.
- iv) In the frozen static port scenario, both groups suspected ice, but the incorrect group abandoned that track early in the search.
- 5) Several general styles of information search strategy can be discerned from the DIG's.
- 6) There is no discernible relationship between the way a pilot collects information for diagnosis and the way he collects information for the destination diversion decision.

The order of scenario presentation was different for a subgroup of the CAT subjects. This was done to test for learning effects within the diagnostic test session.

The only discernible learning effect is with respect to time between inquiries. The first trial in CAT for both orders of presentation has a higher mean time between inquiries than all subsequent trials. The reduction in inquiry time apparently comes with familiarity in using the touch CRT system. The responses of pilots are essentially scenario dependent. Number of inquiries, number of tracks and number of inquiries per track all depend more on the content of the scenario in question than on its order position within test session.

Four subjects were re-tested on scenario two (vacuum pump failure) several months after their first exposure to that scenario. Correctness scores and track performance showed little change. However, the mean and variance of time between inquiries were lower for the second trial.

Although the data samples are small, the evidence indicates that learning in terms of timing of inquiries does take place both within and across scenarios. Other performance measures do not seem to be affected. This suggests that information seeking style is governed more by scenario content than by experience with CAT gleaned from past scenarios.

In general CAT appears to offer the most potential for learning how to attack particular families of CIFE's. In addition it can be used to show pilot deficiencies in understanding the nature of CIFE's.

F. CAT versus Paper and Pencil

If one is careful to account for the differences in experience levels of the two groups of subjects, the results of the CAT experiments are generally consistent with those reported in the 1981 report on paper and pencil testing. Among the few differences noted are the following:

- The P/P group made fewer inquiries and explored fewer tracks in diagnosis than did the CAT group. This suggests that the computer form of testing may be less inhibiting and more user friendly than face to face verbal exchange.
- 2) The CAT group performed a wider search for destination information than did the P/P group.
- 3) Although the most frequently selected airports were different between the groups, the physical characteristics of the most popular airports were similar in both cases.

G. PLATO-GAT Highlights

The PLATO-GAT experiments attempted to produce a GAT simulator experience using a computer terminal, in so far as resource management and information seeking were concerned. Both diagnosis and destination diversion scenarios were combined. This required the subject to diagnose a problem while enroute in a dynamic simulated environment in which alternate destinations could be selected.

General observations about PLATO-GAT include the following:

 Pilots exhibit a wide variety of resource management styles similar to those found in the full mission simulation GAT studies reported in 1981 (6).

- 2) Pilots show a strong preoccupation with problem symptoms and corrections which lead to poor positional awareness and poor choice of destination airports. This characteristic was also noted in the full mission GAT studies.
- 3) Most pilots are reluctant to accept the fact that no diagnosis or solution may be available in flight.
- 4) Pilots do not appear to plan for problems in terms of constantly keeping enroute alternates in mind in case a problem does develop. Their management style is more reactive than pre-structured.

H. Training Potential

The computer aided testing instruments described in this report were developed as research tools to be used to better understand the decision making styles of pilots faced with critical in-flight events. However, from observations by the research team and from repeated comments by subject pilots, these tools may have even greater potential for pilot training.

CAT offers the possibility of self-paced instruction in acquiring basic aeronautical knowledge through an expanded version of the current knowledge survey. Such an expansion could easily include immediate feedback on the correctness of an answer, reasons for the correct response and references on availability of additional information. CAT also offers the opportunity to experience a variety of critical inflight events from the safety of the computer terminal. In addition to a wealth of potential simulated decision experiences, CAT could be used to uncover pilot deficiencies in their current understanding of the nature of CIFE's and to train pilots to develop more efficient information search habits.

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