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LANDSAT-D Investigations Workshop

May 13-14, 1982

Goddard Space Flight Center



Day 1



(E83-10241) LANDSAT-D INVESTIGATIONS
WORKSHOP (NASA) 266 p HC A12/MF A01

CSCL 05A

N83-21482

Unclas

G3/43 00241

E83-10241
TM-85274

N83-21482 #

Agenda

John Barker

Agenda

LANDSAT-D INVESTIGATIONS WORKSHOP - BLDG. 26, ROOM 205

Thursday, 13 May 1982

8:30 am	Agenda	Barker
8:35 am	Welcome, Science Organization and Introduction of Investigators	Salomonson
9:00 am	Landsat-D Project Status	Busse
9:15 am	Landsat-D Ground Segment	Webb
10:15 am	BREAK (Photo Session)	
10:45 am	Early Access TM Processing	Lyon
11:00 am	Landsat-D Data Acquisition and Availability	Freden
11:30 am	Landsat-D Performance Characterization	Barker
12:00 pm	Introduction of Technical Experts and Science Representatives	
12:30 pm	Lunch and Informal Investigations Team Interaction	

Thursday, 13 May 1982 (Cont.)

- | | | |
|---------|---|--------|
| 2:00 pm | Introduction to MSS Pre-NOAA
Characterization | Alford |
| 2:15 pm | MSS Radiometric Sensor
Performance <ul style="list-style-type: none">● Spectral Information● Absolute Calibration● Ground Processing | Barker |
| 3:30 pm | MSS Geometric Sensor
Performance | Banks |
| 4:00 pm | MSS Geometric Processing and
Calibration | Brooks |
| 5:00 pm | Closing Remarks | Barker |
| 5:30 pm | Dinner and Informal Investigations
Team Interaction | |

Thursday, 13 May 1982 (Cont.)

TOURS, etc., Building 28

(Five Tours/Presentations Offered Each Half Hour)

8:30 pm	}	1. Landsat Assessment System (LAS)	Lyon/Fischel
9:00 pm		2. Image Generation Facility (IGF)	GE
9:30 pm	}	3. MSS and TM Sensored Pictures	Barker
10:00 pm		4. Control and Simulation Facility (CSF)	GE
		5. End-to-End System Analysis Study Highlights	Billingsley

Agenda

LANDSAT-D INVESTIGATIONS WORKSHOP

Friday, 14 May 1982

- 8:00 am Informal Investigations Team
Interaction
- 8:30 am Introduction to TM Characterization Barker
- TM Radiometric Sensor Performance
- Spectral Information
 - Absolute Calibration
 - Ground Processing
- 10:15 am BREAK
- 10:30 am TM Geometric Sensor Performance Engel
- 11:30 am TM Geometric Processing —
Flight Segment Beyer
- 12:30 pm Lunch and Informal
Investigations Team Interaction

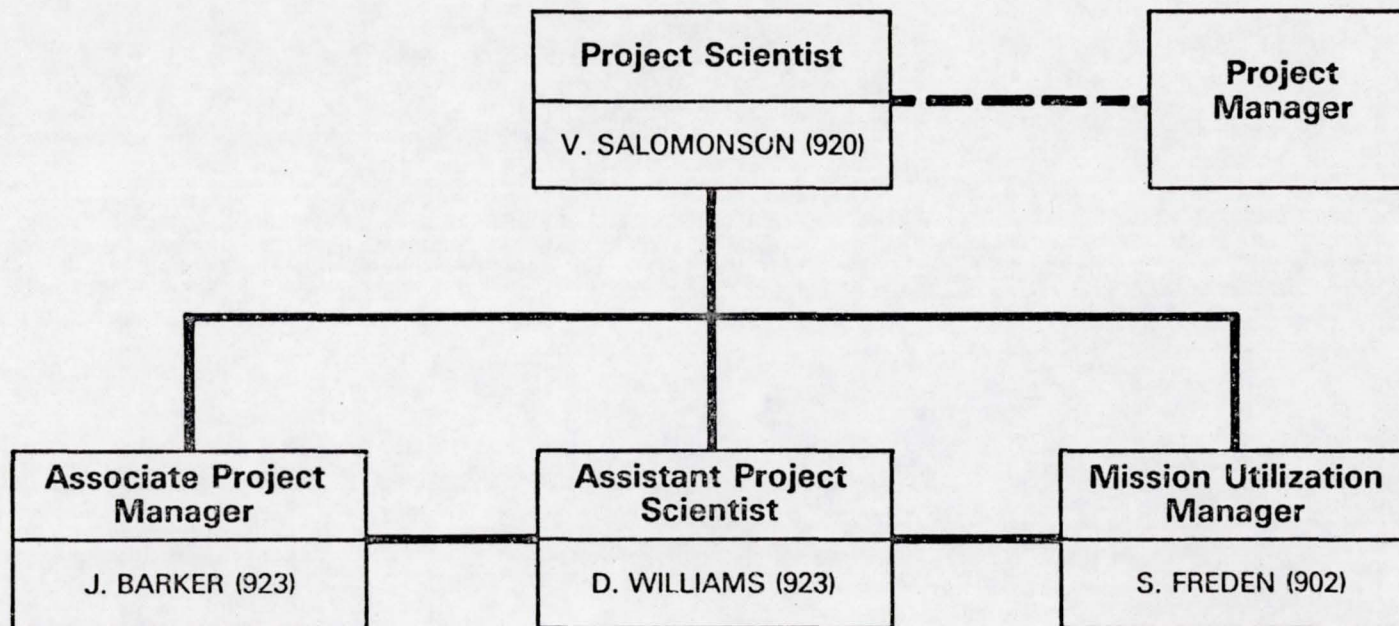
Friday, 14 May 1982 (Cont.)

- | | | |
|---------|---|-----------------|
| 1:45 pm | TM Geometric Processing —
Ground Segment | Beyer |
| 3:00 pm | Early Access TM Processing | Fischel |
| 3:45 pm | Wrap-Up Panel Discussion | Science
Team |
| 4:15 pm | Informal Investigations
Team Interaction | |

**Welcome,
Science Organization
and
Introduction of Investigators**

Vince Salomonson

Science Office Organization



Distributed Responsibilities

PROJECT SCIENTIST

- Overall Management and Direction of Project Science Activities
- Representation of Science Objectives and Activities to NASA Headquarters, GSFC and the User Community
 - Chairman of Landsat-D Technical Users Working Group

ASSOCIATE PROJECT SCIENTIST

- Day to Day Representation to the Project
- Systems Performance for Flight and Ground Segments
- Systems Contractor Management (Santa Barbara Research Center (SBRC), GE) for Science Contracts

Distributed Responsibilities (Cont.)

ASSISTANT PROJECT SCIENTIST

- Science Monitor
- Science Resources
 - Manpower
 - Dollars

MISSION UTILIZATION MANAGER

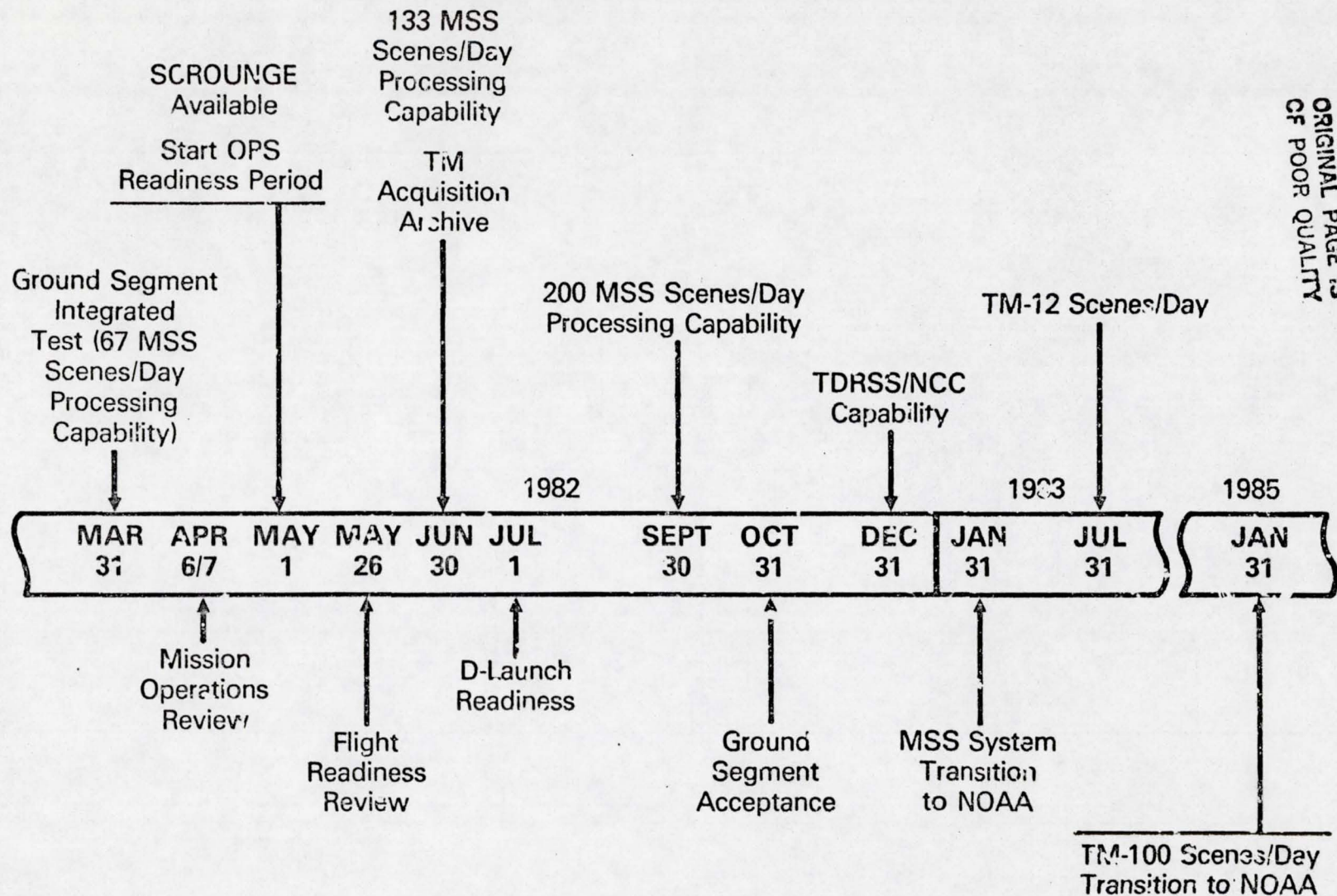
- Scene/Data Selection
 - Acquisition
 - Processing
 - Distribution
- Out-of House Investigators Management
 - Contracts
 - Grants
 - MOU's
 - Scientific Monitors
 - Contract Monitors

All Responsibilities Require Close Contact and Frequent Communication with All Elements of the Project: e.g. LAS, ADDS, Software Manager, Mission Operations Manager, etc.

Landsat-D Project Status

Jon Busse

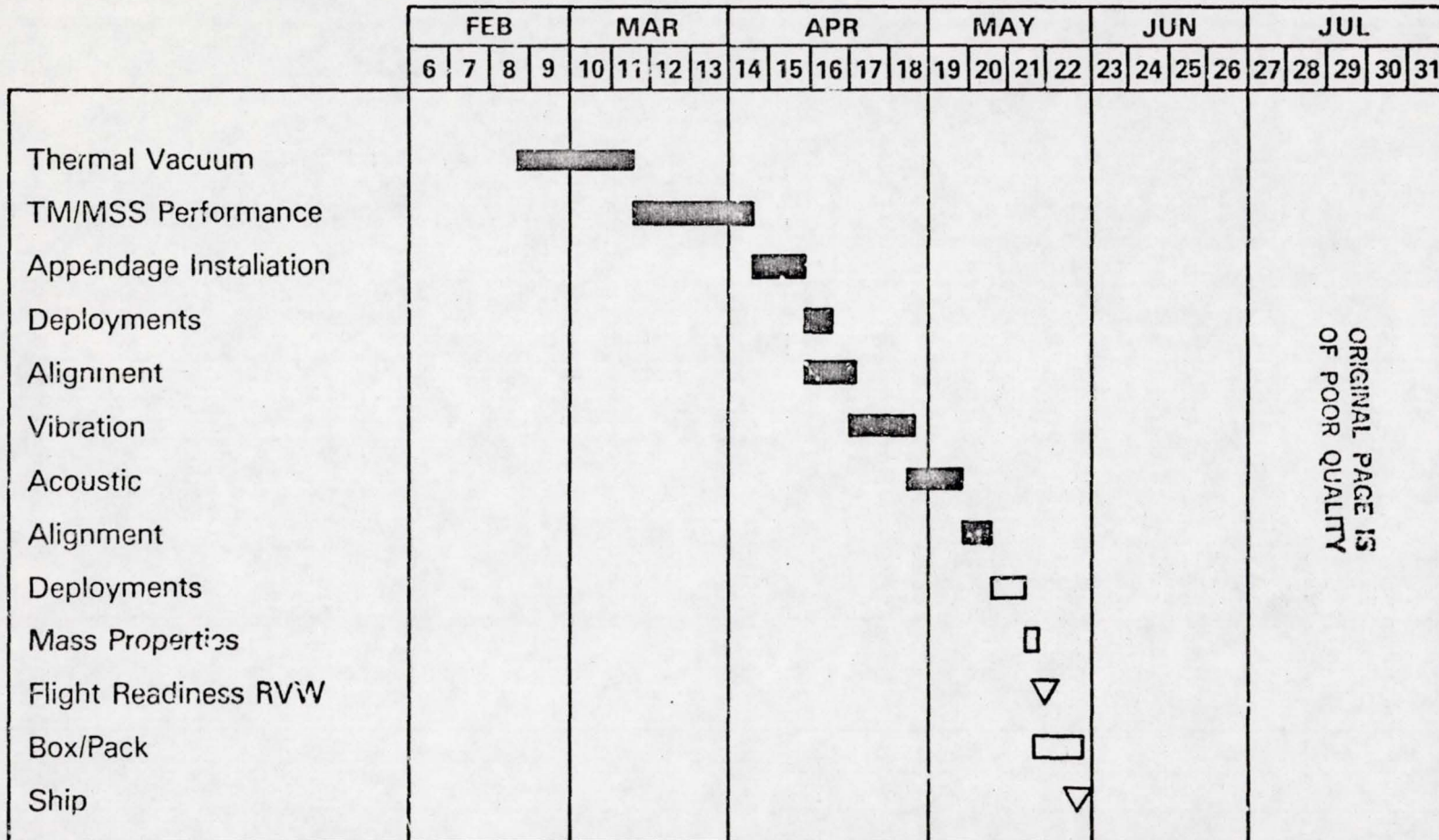
Landsat-D Key Events



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Flight Segment Status

COMPLETED THERMAL VACUUM TEST MARCH 11, 1982



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Landsat-D Ground Segment

Bill Webb

Landsat-D System Overview

- System Requirements
- Flight Segment
- Ground Segment

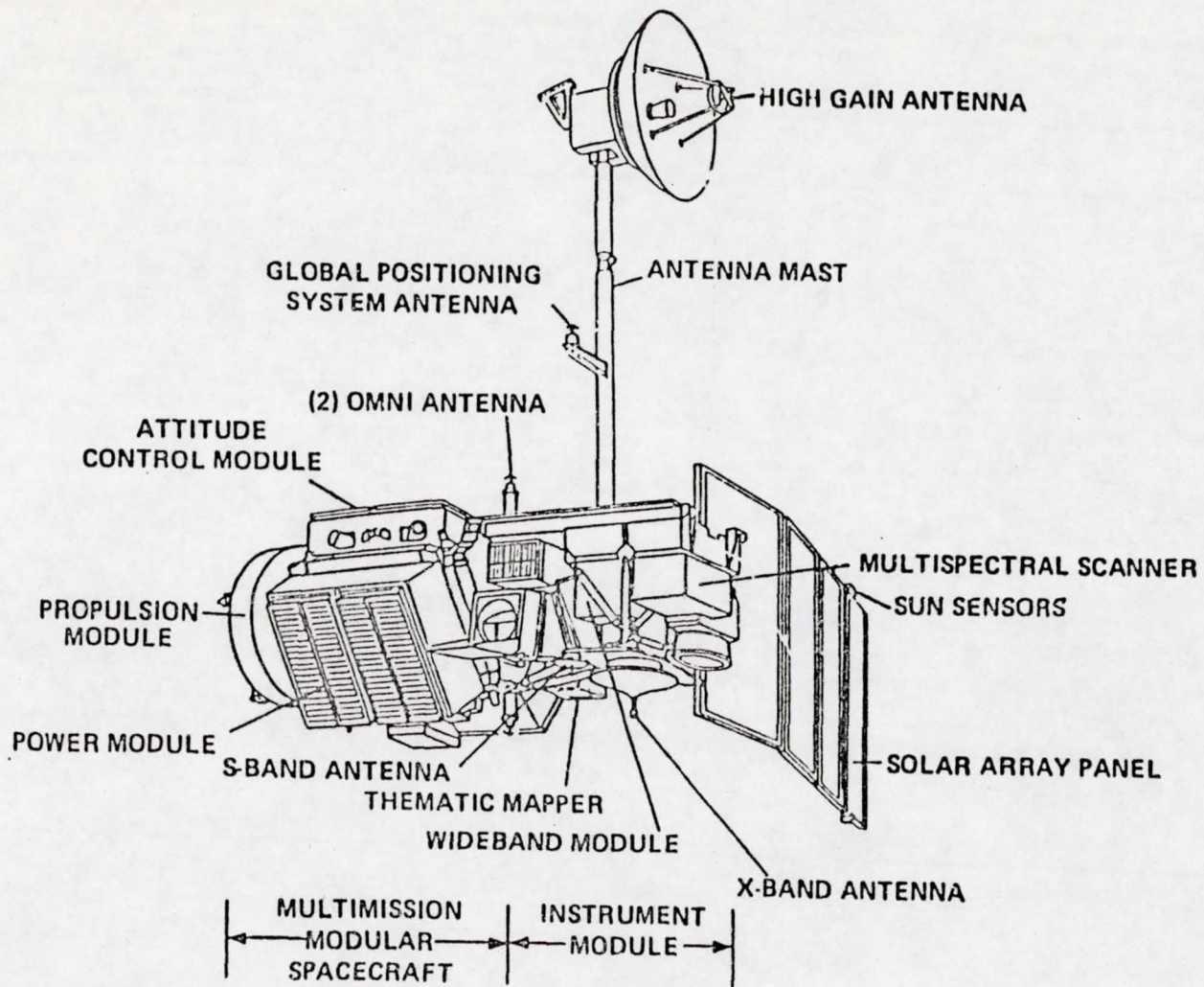
Landsat-D System Requirements

- **Orbit**
 - Altitude: 705.3 km
 - Descending Node: 9:30-10:00 a.m.
 - **Launch Vehicle**
 - Delta 3920
 - **Instruments**
 - Multispectral Scanner
 - 4 Band
 - 80 Meter IFOV
 - Thematic Mapper
 - 7 Band
 - 30 Meter IFOV
 - **Flight Segment**
 - Uses Multimission Modular Spacecraft
 - Shuttle Retrievable
 - **Mission Life**
 - 3 Years
 - **1 Spacecraft Operation at a Time**
 - **Ephemeris Data**
 - Global Positioning System
 - Predicted/Uplink
 - **Coverage**
 - Ground Station Tracking Data Network Initially
 - Tracking and Data Relay Satellite System When Available
 - **Scenes/Day — Flight Segment**
 - NASA (200 MSS; 100 TM)
 - Foreign (337 MSS; 150 TM)
 - **Data Quantity — Ground Segment**
 - MSS
 - Archival Data — 200 Scenes/Day
 - TM
 - TM Evaluation— Begins July 30, 1982
1 Scene/Day
 - TM R&D— Begins July 30, 1983
Archive Data— 12 Scene/Day
Fully Processed— 12 Scene/Day
 - TM Operational— Begins January 31, 1985
Archive Data— 100 Scene/Day
Fully Processed— 50 Scene/Day Film/HDT
10 Scene/Day CCT
- Products Through Ground Segment Within 48 Hours**
- High Density Tapes
 - Computer Compatible Tapes
 - TM Film
- **Data Quality**
 - Geodetic Accuracy: 0.5 Pixel, 90% of the Time
 - Temporal Registration: 0.3 Pixel, 90% of the Time
 - Radiometric Correction to ± 1 Quantum Level

**LANDSAT-D
FLIGHT SEGMENT**

1-17

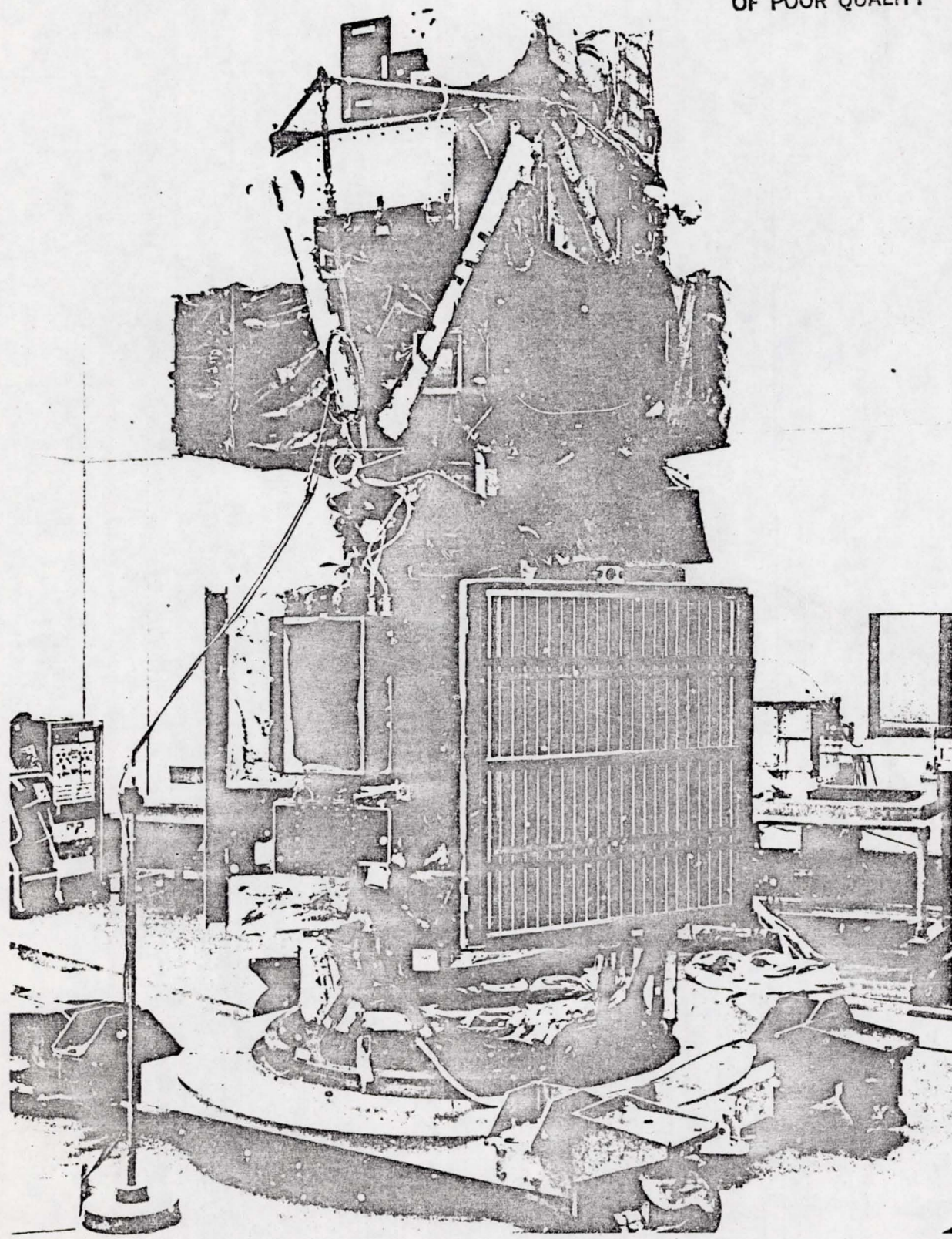
LANDSAT D FLIGHT SEGMENT



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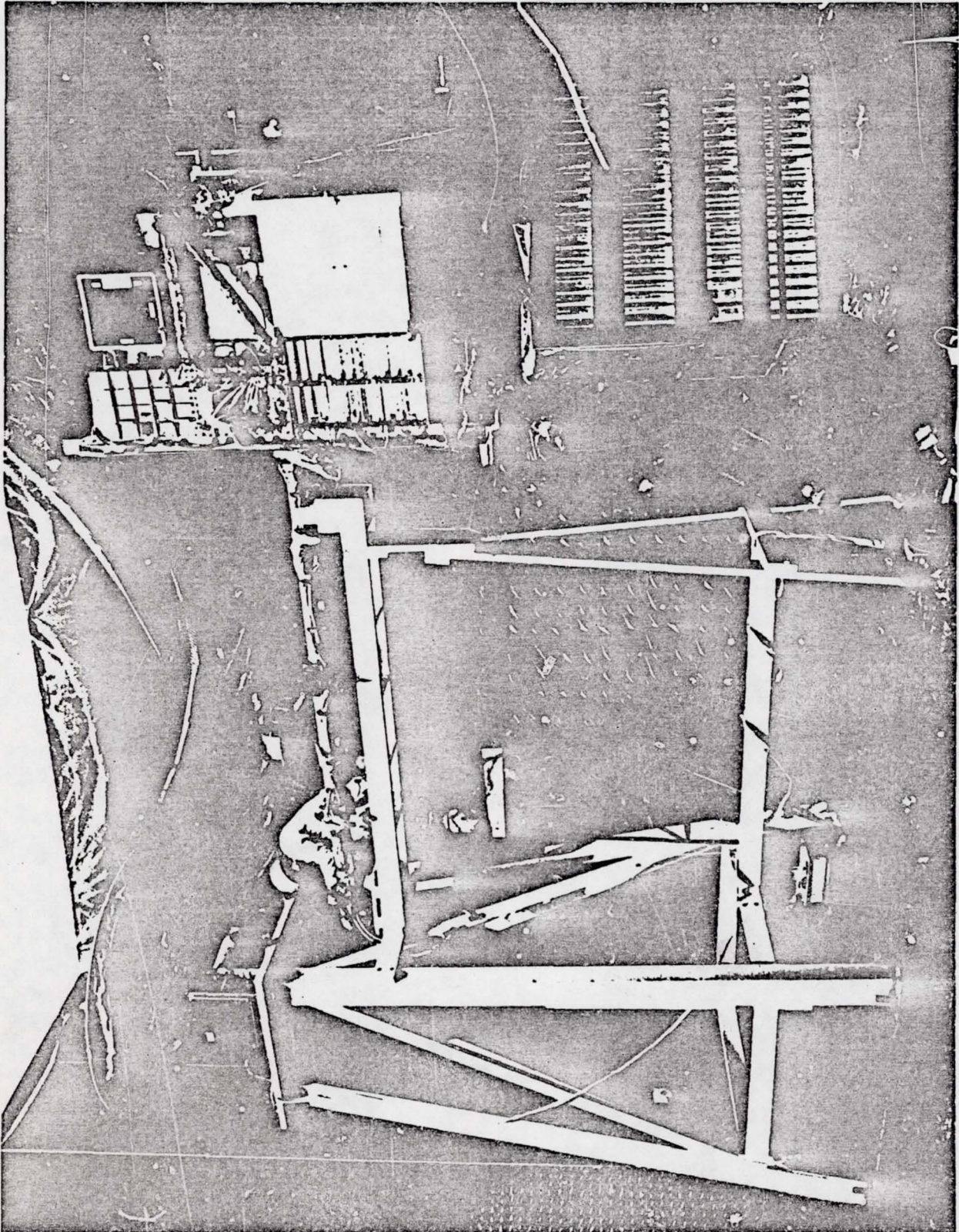
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1-19

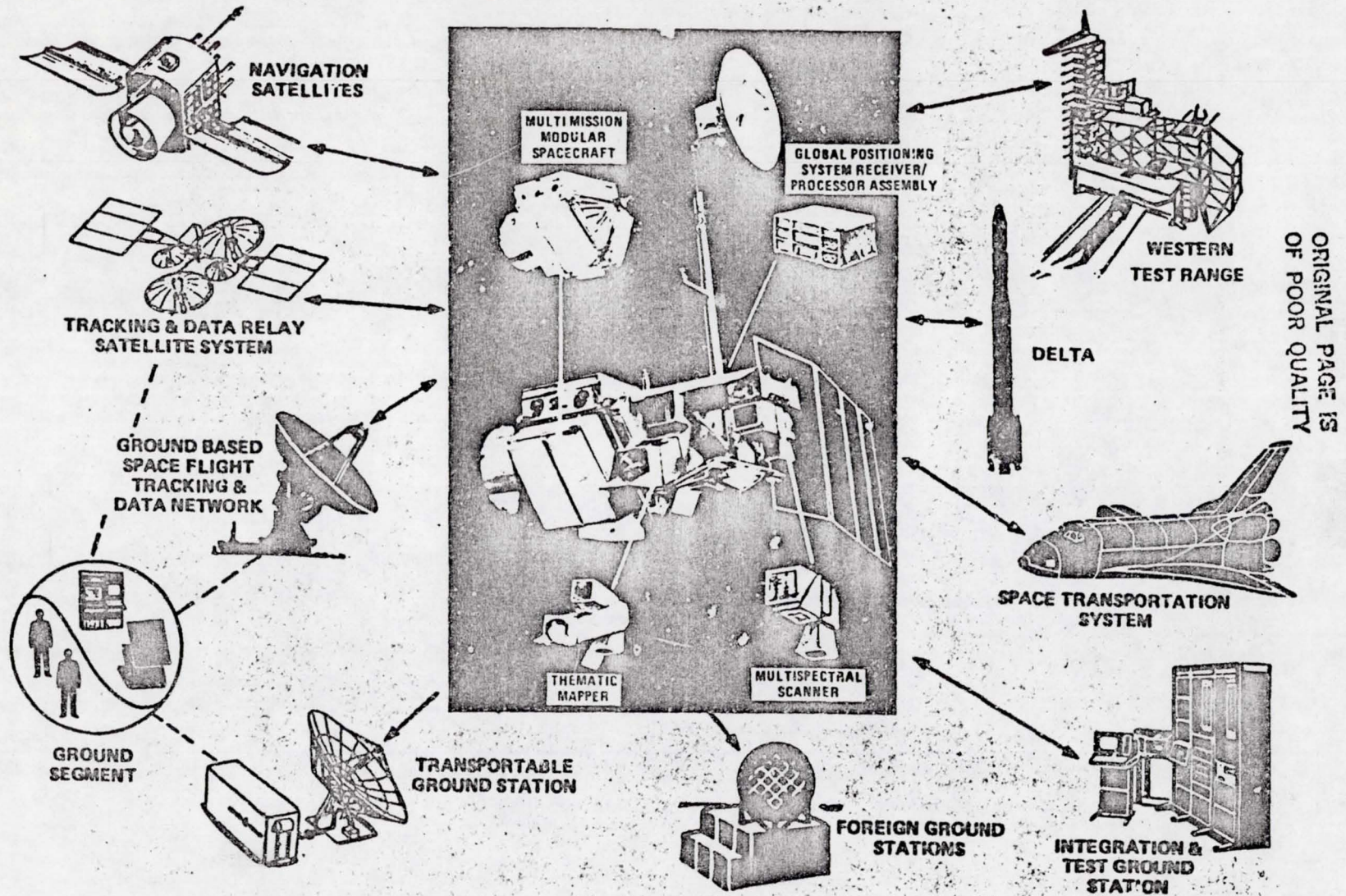


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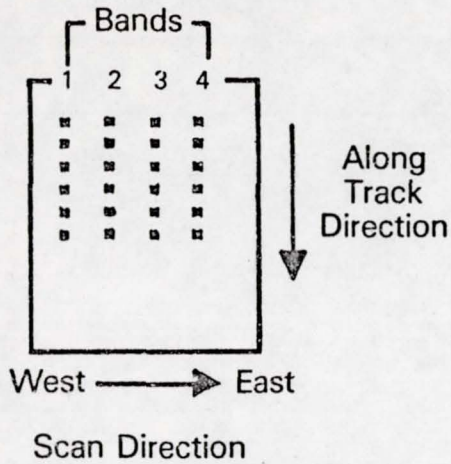


FLIGHT SEGMENT INTERFACES



Multispectral Scanner (MSS) Sensor

MSS Detector Details



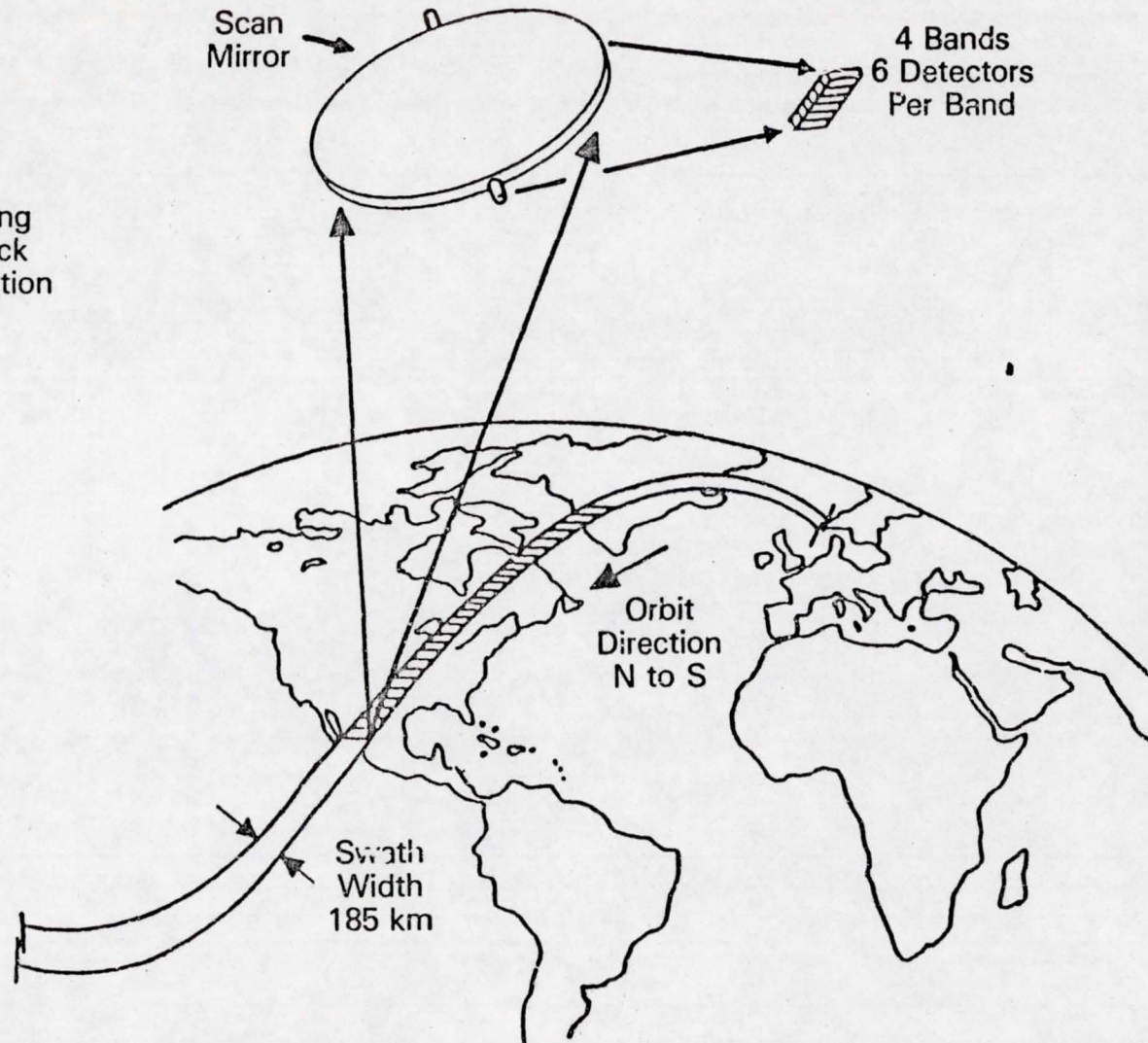
Band	Spectral Ranges μm
1	.5 - .6
2	.6 - .7
3	.7 - .8
4	.8 - 1.1

Ground IFOV

All Bands - 83 Meters

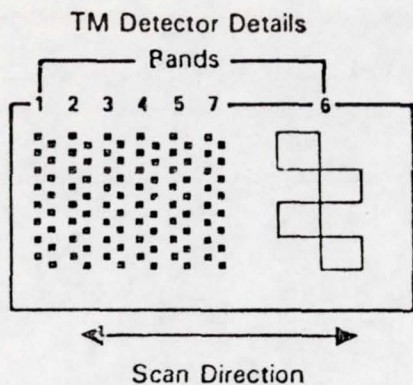
Data Rate - 15.06 Mbps

Quantization Levels - 64



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THEMATIC MAPPER (TM) SENSOR



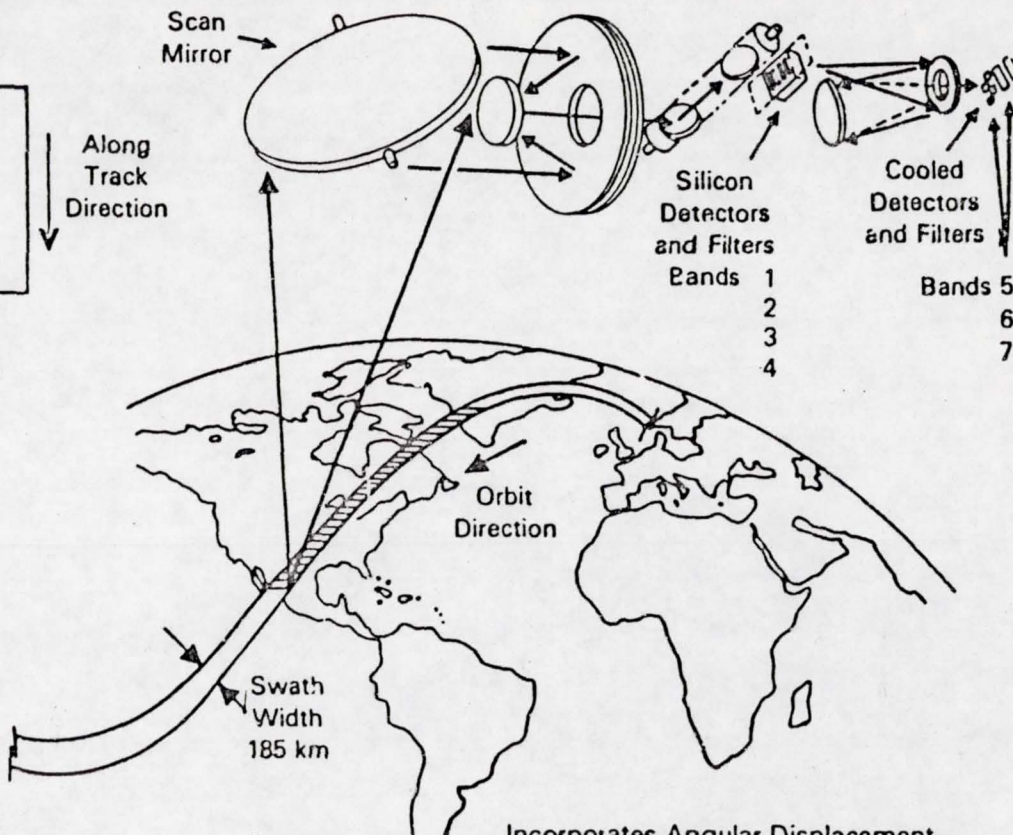
BAND	SPECTRAL RANGES μm
1	45 - 52
2	52 - 60
3	63 - 69
4	76 - 90
5	1.55 - 1.75
6	10.4 - 12.5
7	2.08 - 2.35

Ground IFOV

Bands 1-5 & 7 30 Meters
Band 6 120 Meters

Data Rate - 04.9 Mbps

Quantization Levels - 256

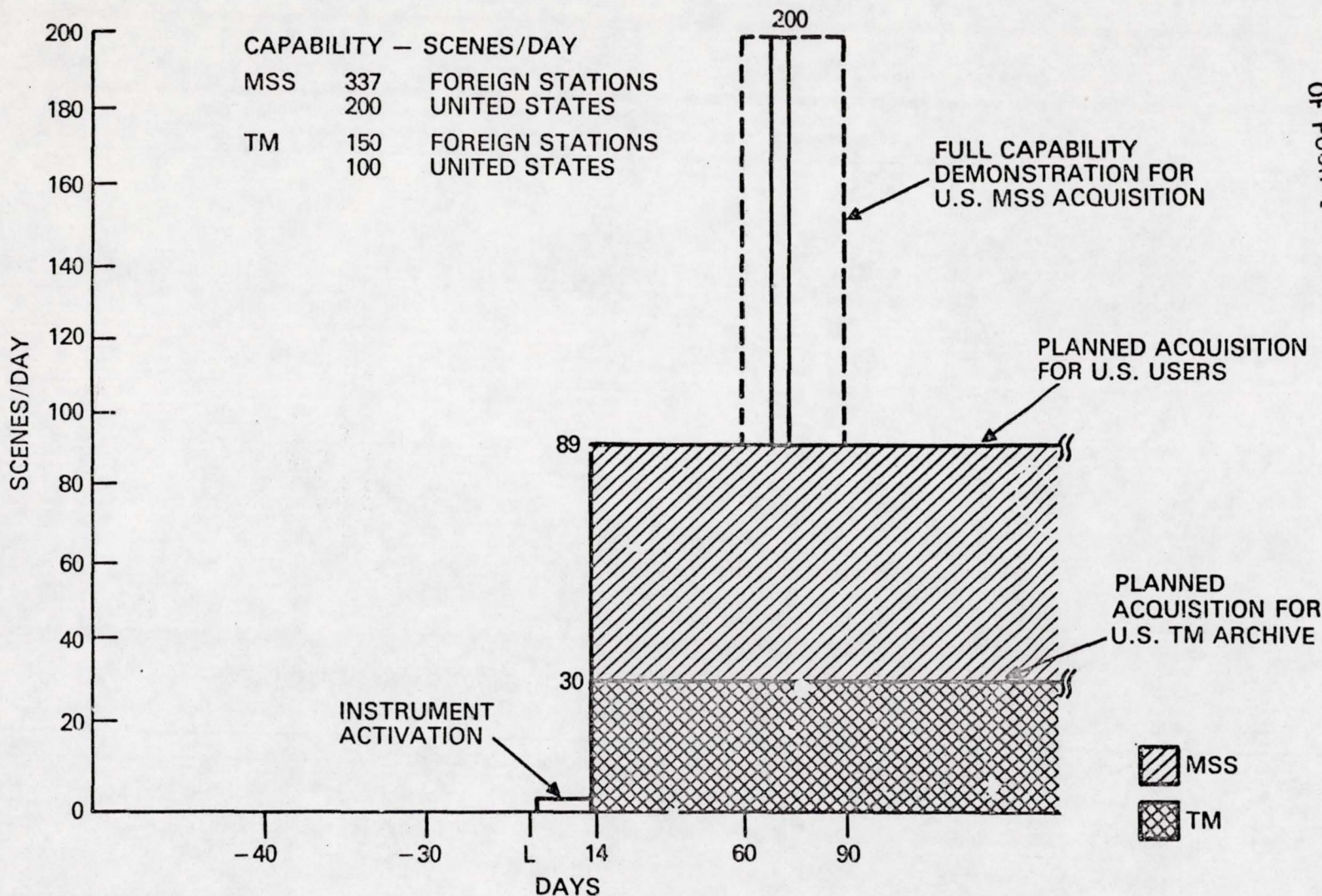


Incorporates Angular Displacement Sensor (ADS)

Measures Angle of Motion from 2 to 125 Hz
Source of Jitter Compensation Data

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Acquisition

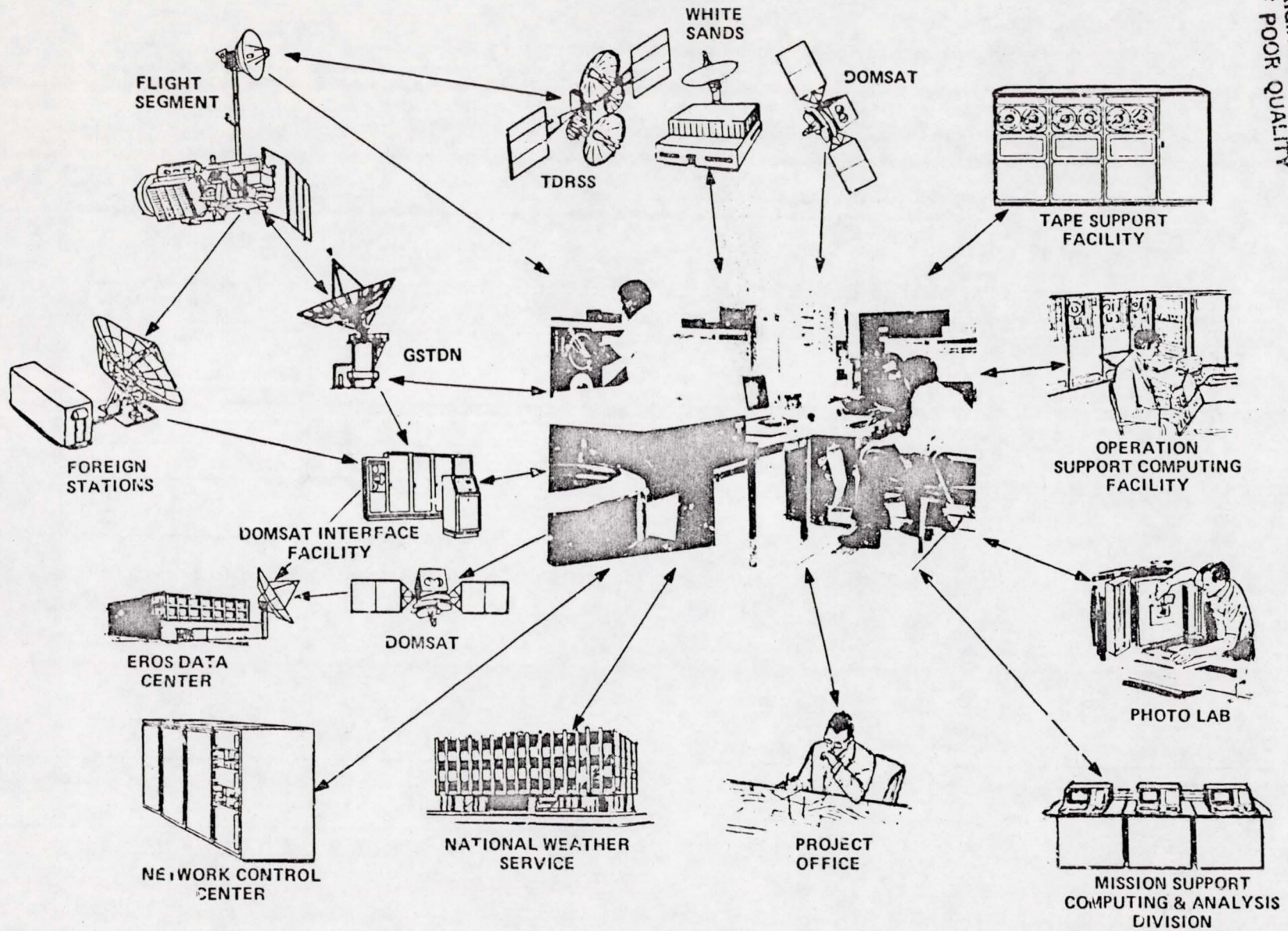


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**LANDSAT-D
GROUND SEGMENT**

1-25

GROUND SEGMENT INTERFACES

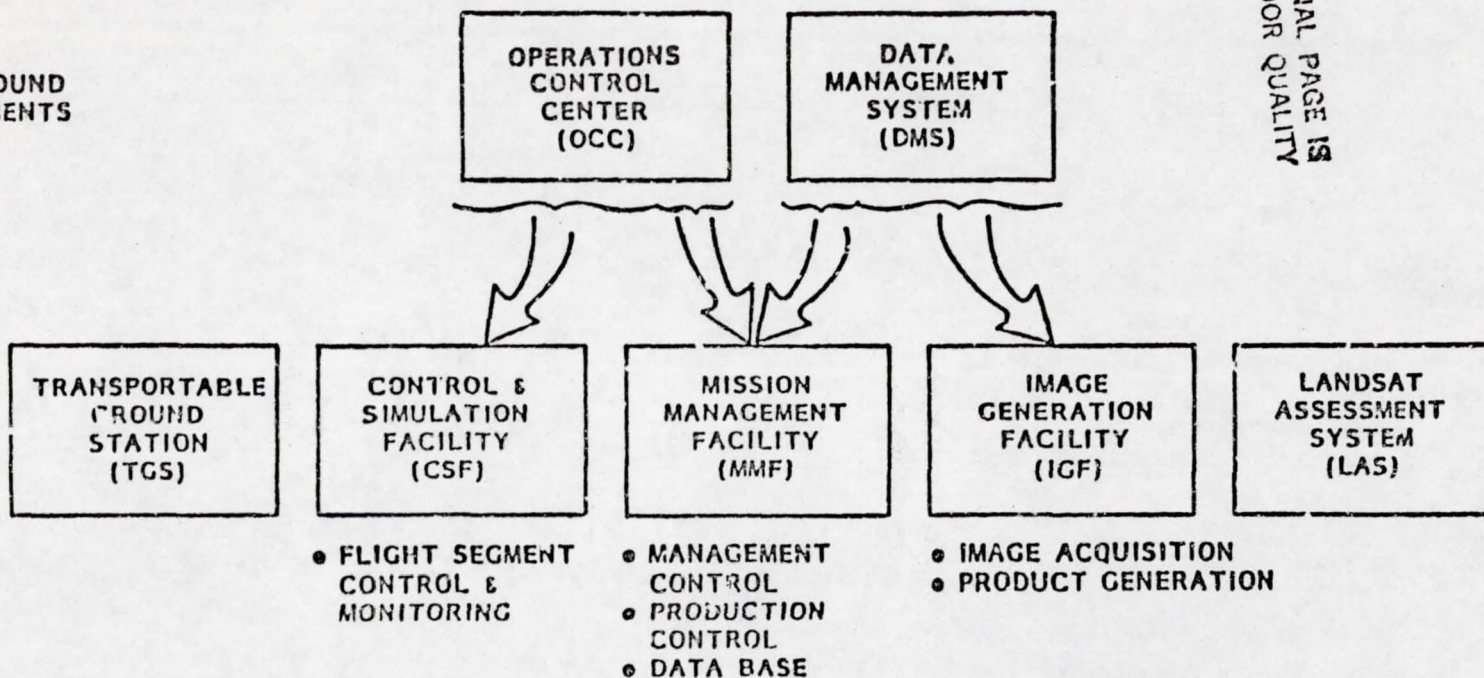


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PARTITIONING THE GROUND SEGMENT

TYPICAL GROUND
SYSTEM ELEMENTS

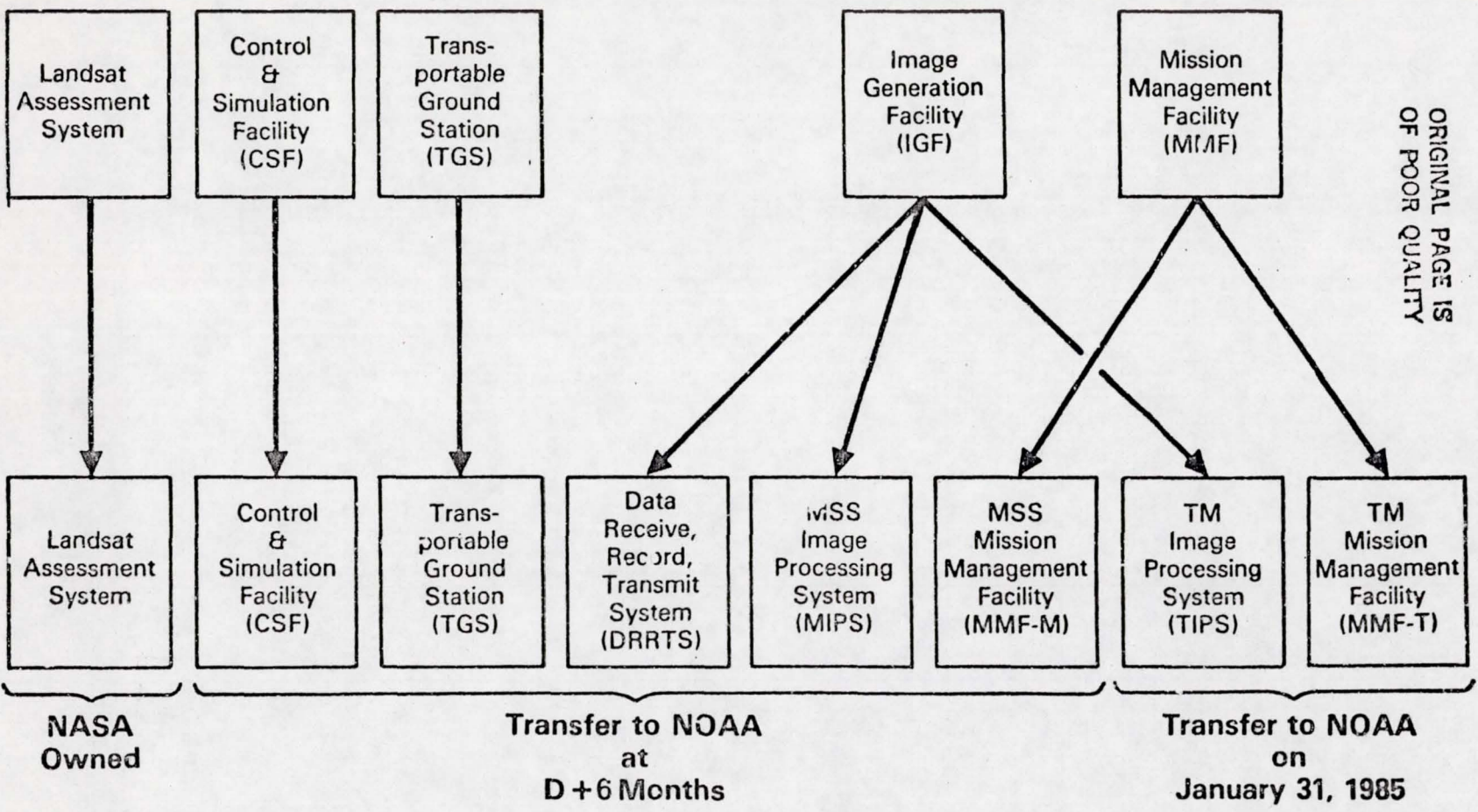
LANDSAT-D
ELEMENTS



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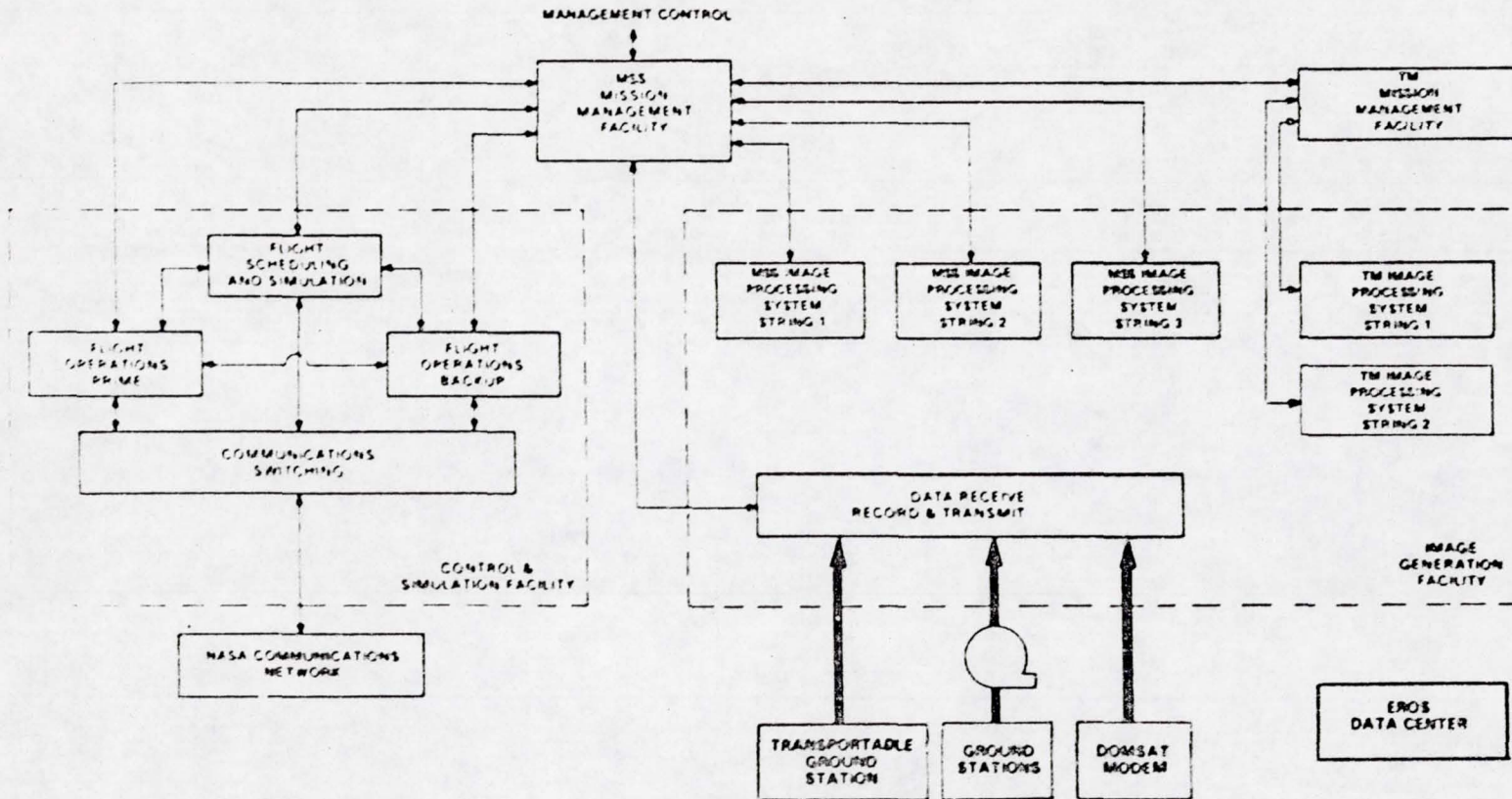
Mission Management Facility and Image Generation Facility Partitioning

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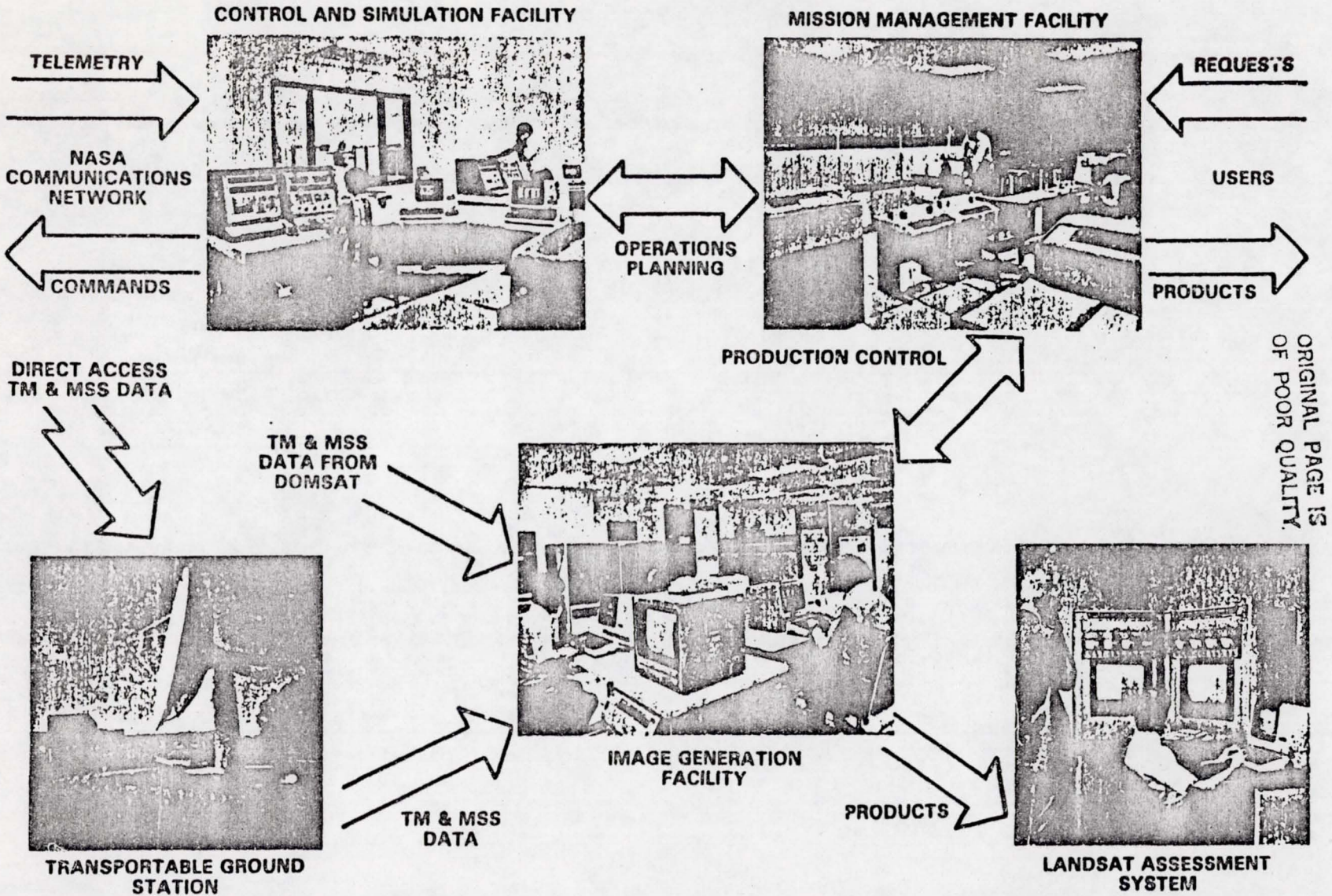


LANDSAT-D GROUND SEGMENT

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LANDSAT D GROUND SEGMENT



Mission Management Facility

Major Functions

- **Data Acquisition**
 - **PROCESS REQUESTS** for Data Acquisition
 - **PROVIDE CANDIDATE** Scene Data **ACQUISITION LISTS** for Satellite Operations Planning and Scheduling
 - **ACCOUNT FOR TELEMETRY** Data Acquisition
 - **ACCOUNT FOR IMAGE** Data Acquisition
- **Data Archive**
 - **SCHEDULE** Archival Processing
 - **MAINTAIN** Archival **DATA BASE** and Produce Image Catalogs
- **Product Generation**
 - **PROCESS REQUESTS** for Product Generation
 - **SCHEDULE** Product Generation
- **Ground Segment Management**
 - Maintain Ground Segment **SUPPLIES INVENTORY**
 - **TRACK** Ground Segment **PROBLEMS**
 - Provide **VERIFICATION AND SELF TEST** Capability
 - Provide **MANAGEMENT REPORTS**

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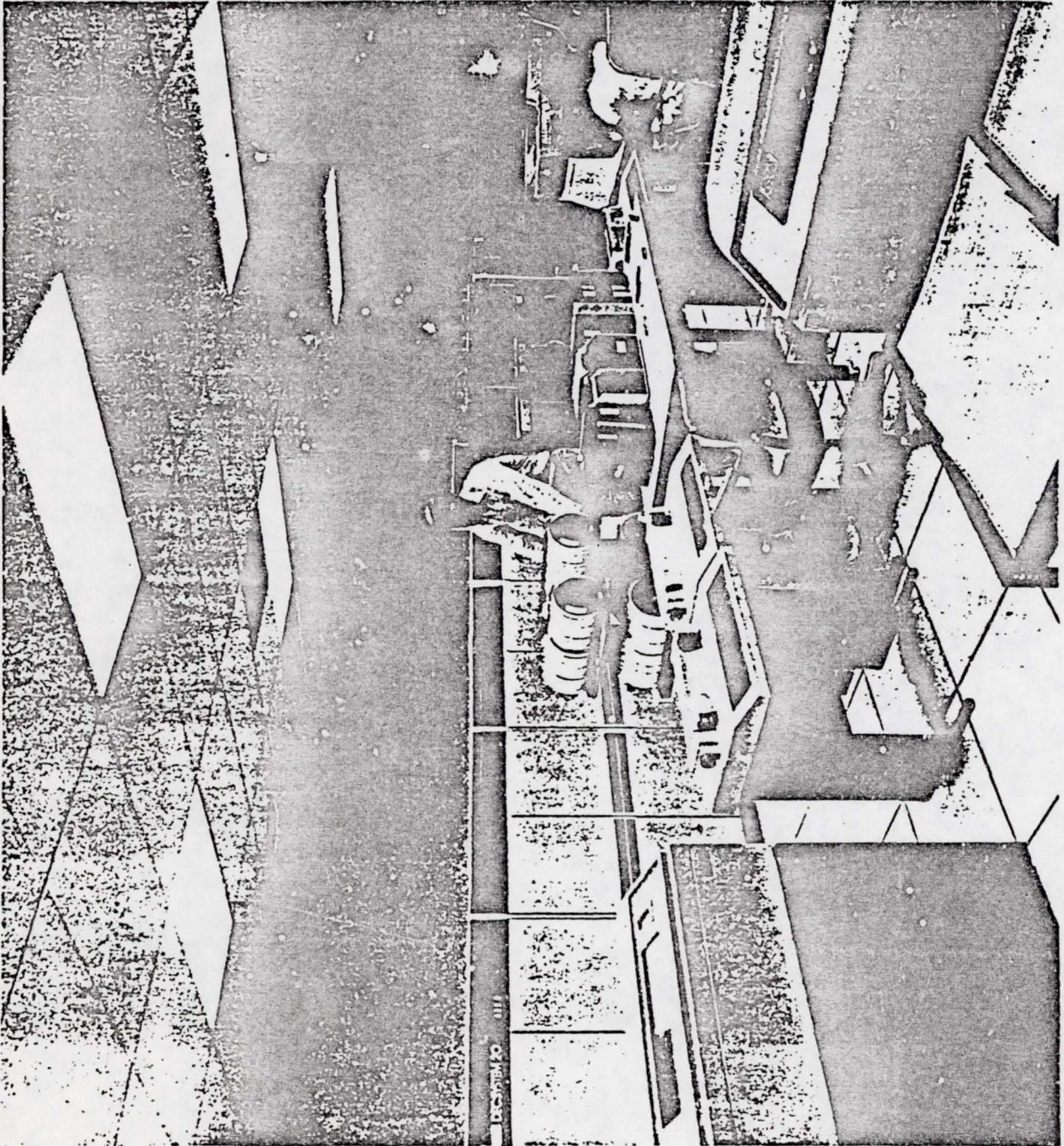


IMAGE GENERATION FACILITY

- **DATA RECEIVE, RECORD TRANSMIT SYSTEM**
- **MSS IMAGE PROCESSING SYSTEM**
- **TM IMAGE PROCESSING SYSTEM**

DATA RECEIVE, RECORD, TRANSMIT SYSTEM MAJOR FUNCTIONS

- **RECORD MSS AND TM DATA**
- **GENERATE TAPE DIRECTORIES**
- **COPY HIGH DENSITY TAPES**
- **ARCHIVE HIGH DENSITY TAPES**
- **TRANSMIT MSS DATA TO EROS DATA CENTER**

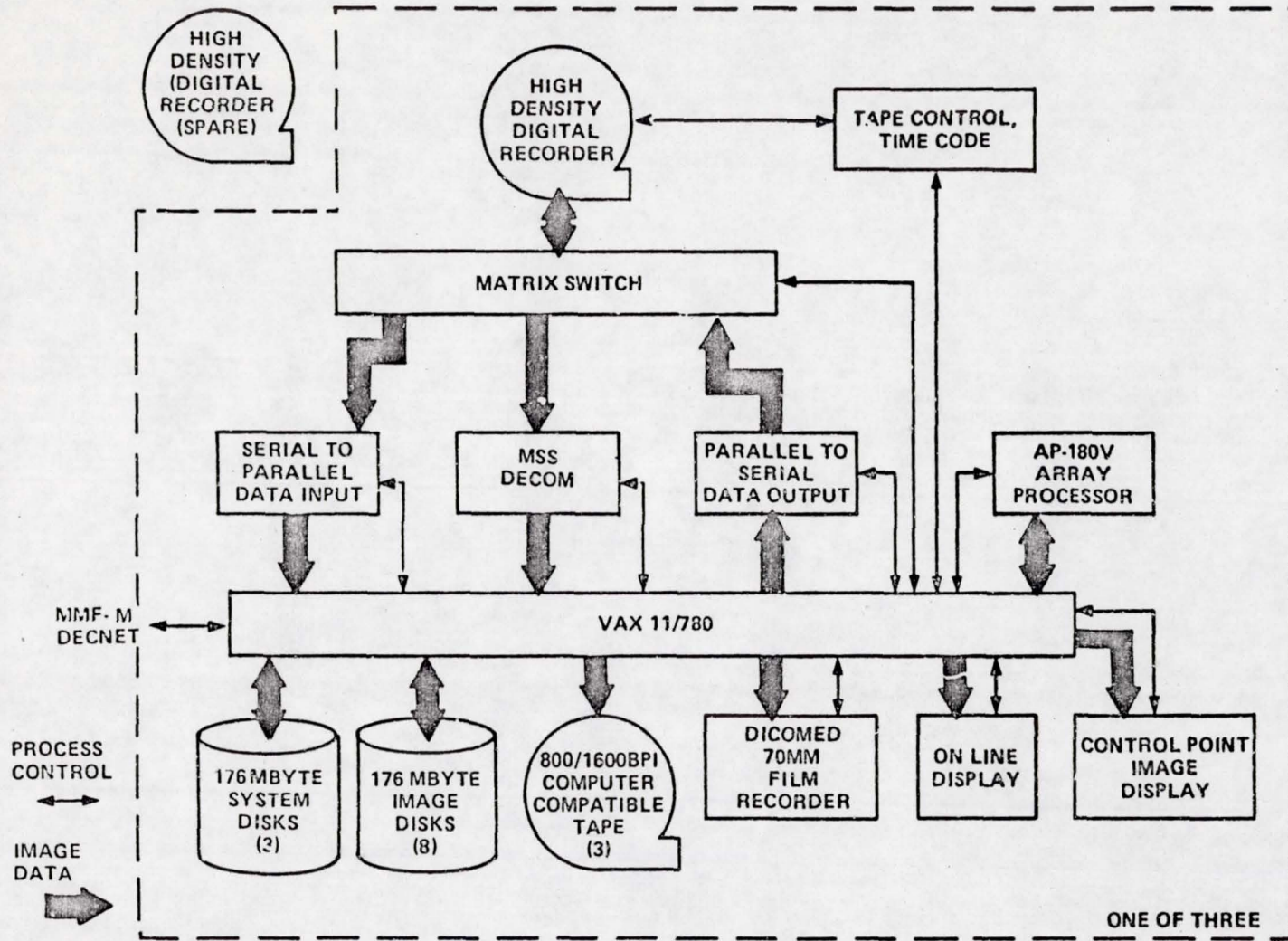
MSS IMAGE PROCESSING SYSTEM MAJOR FUNCTIONS

- **PROCESS IMAGE RELATED TELEMETRY
(ATTITUDE AND EPHEMERIS)**

- **GENERATE 28-TRACK ARCHIVAL TAPES**
 - **RADIOMETRICALLY CORRECTED**
 - **GEOMETRIC CORRECTIONS APPENDED**

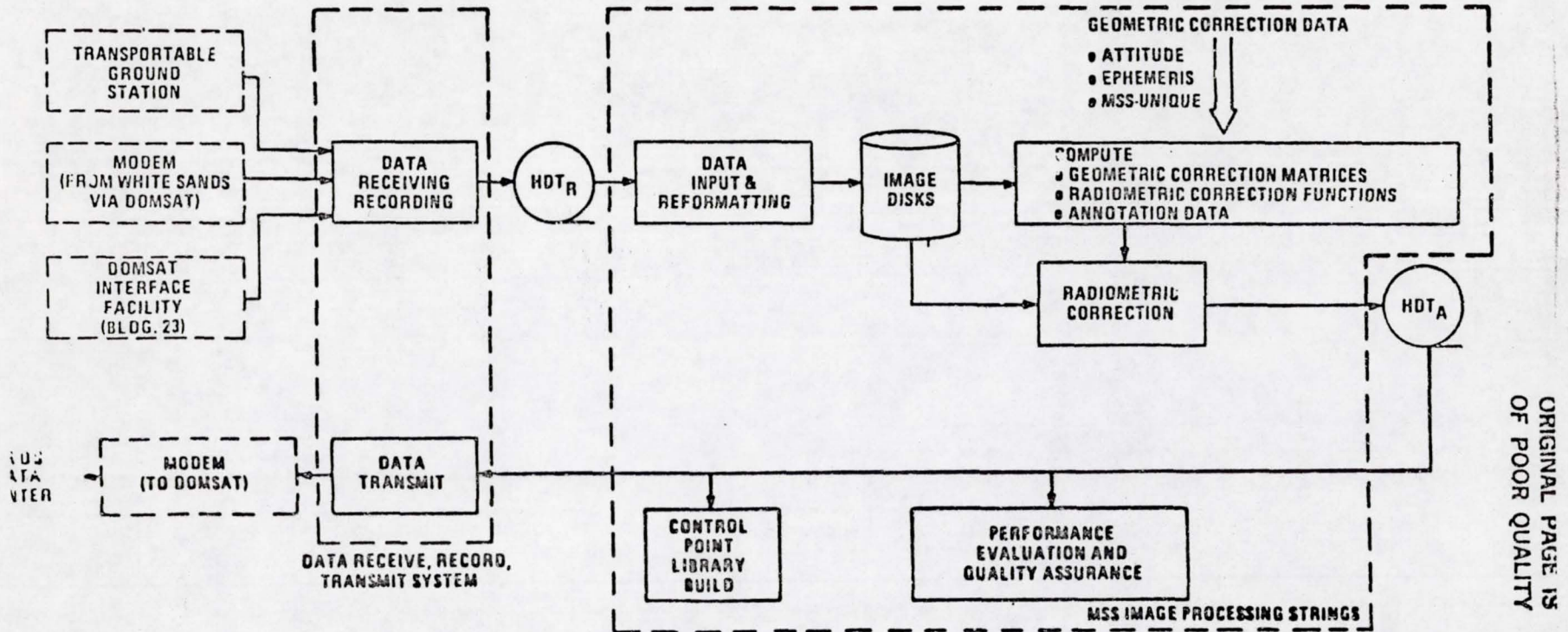
- **CREATE QUALITY ASSURANCE PRODUCTS**
 - **COMPUTER COMPATIBLE TAPES**
 - **REPORTS**
 - **FULLY CORRECTED HIGH RESOLUTION FILM**

MSS IMAGE PROCESSING SYSTEM HARDWARE



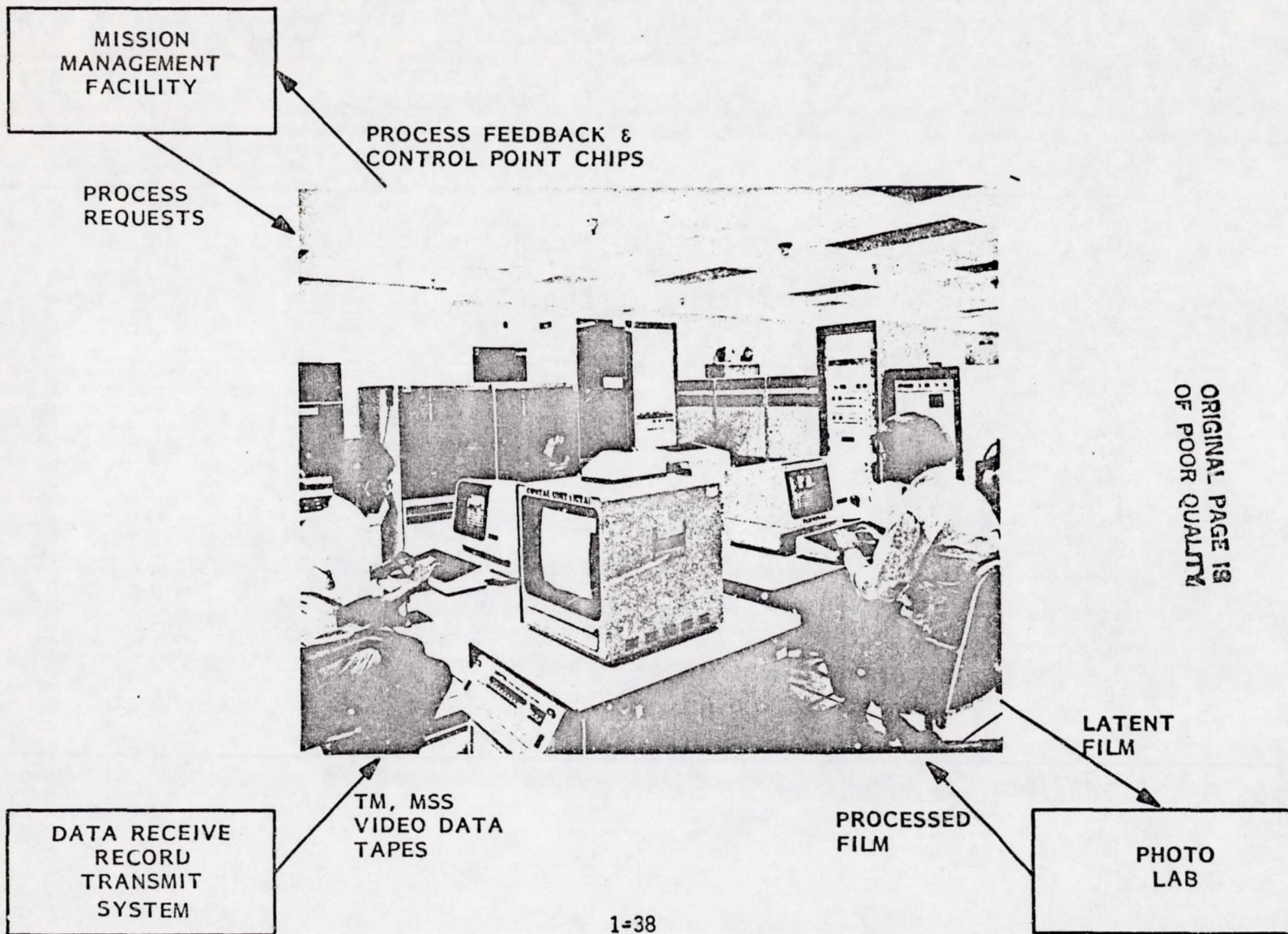
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MULTISPECTRAL SCANNER IMAGE GENERATION PROCESS FLOW



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MSS IMAGE PROCESSING SYSTEM INTERFACES



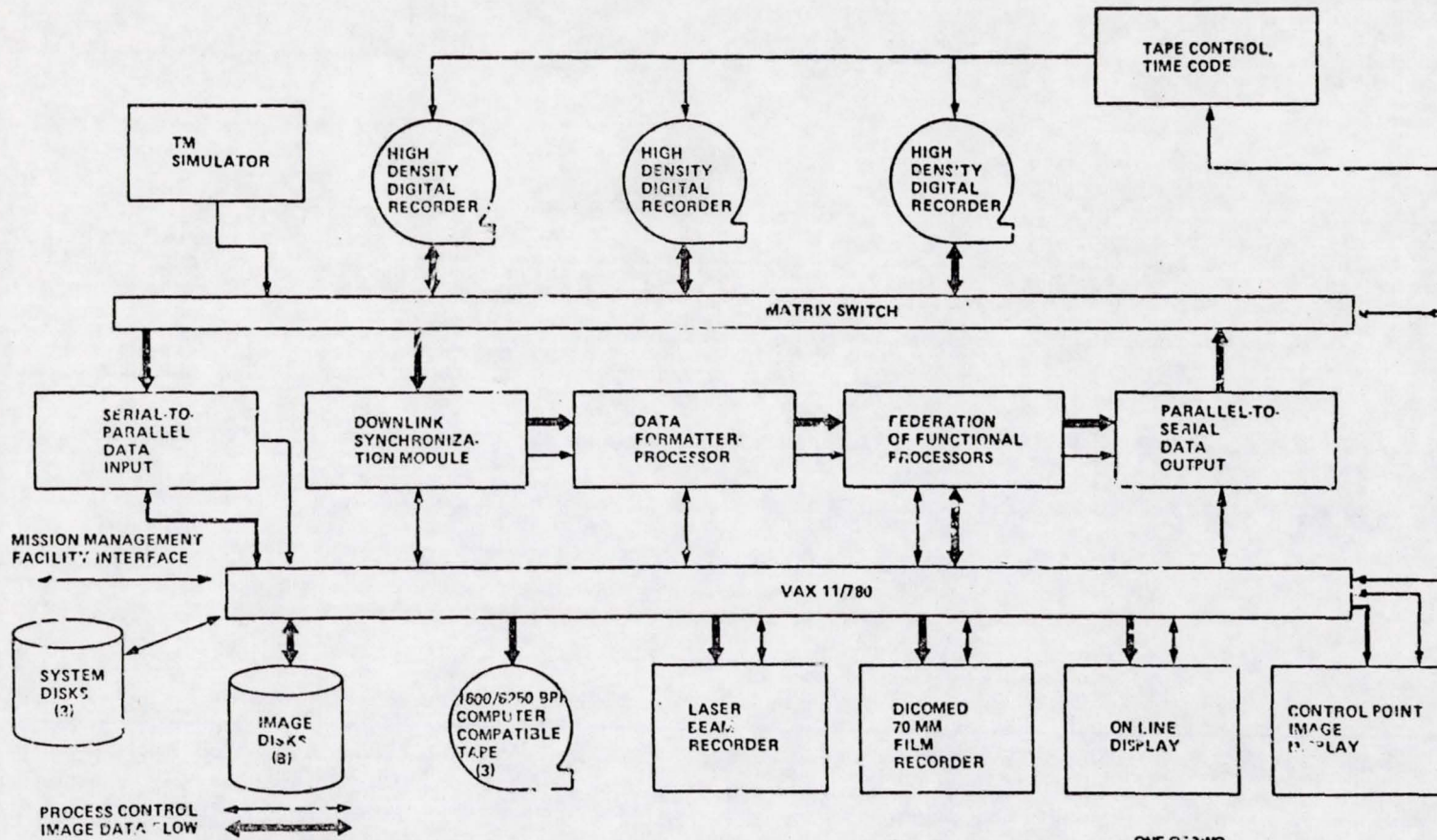
TM IMAGE PROCESSING SYSTEM MAJOR FUNCTIONS

- PROCESS IMAGE RELATED CORRECTION DATA
 - GYRO
 - ANGULAR DISPLACEMENT SENSOR
 - MIRROR SCAN CORRECTION
 - EPHEMERIS
- GENERATE 28-TRACK ARCHIVAL TAPES
 - RADIOMETRICALLY CORRECTED
 - GEOMETRIC CORRECTIONS APPENDED
- GENERATE 28-TRACK FULLY CORRECTED TAPE
- GENERATE HIGH RESOLUTION FILM AND COMPUTER COMPATIBLE TAPE PRODUCTS
- CREATE QUALITY ASSURANCE REPORTS

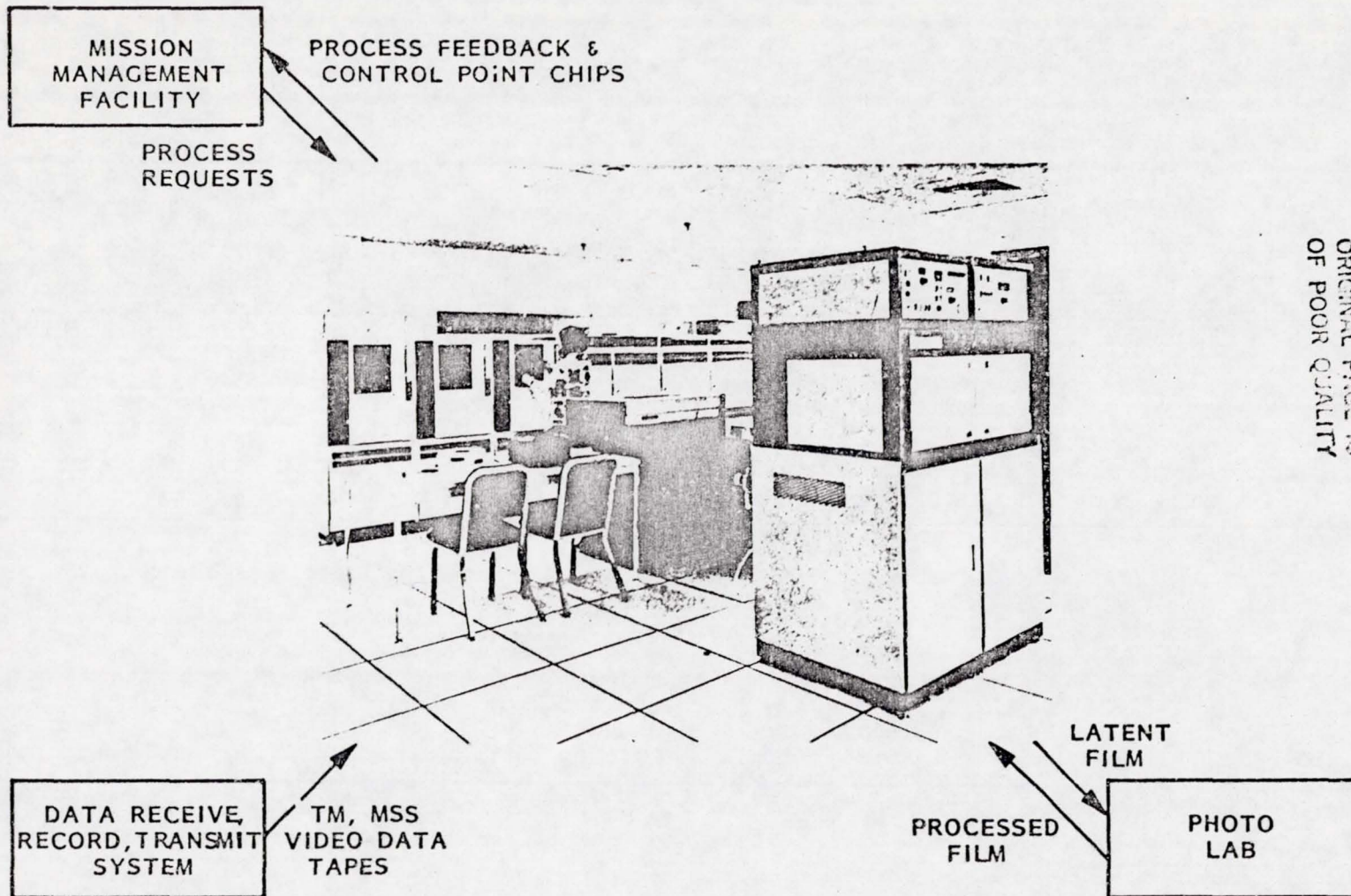
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TM IMAGE PROCESSING SYSTEM HARDWARE

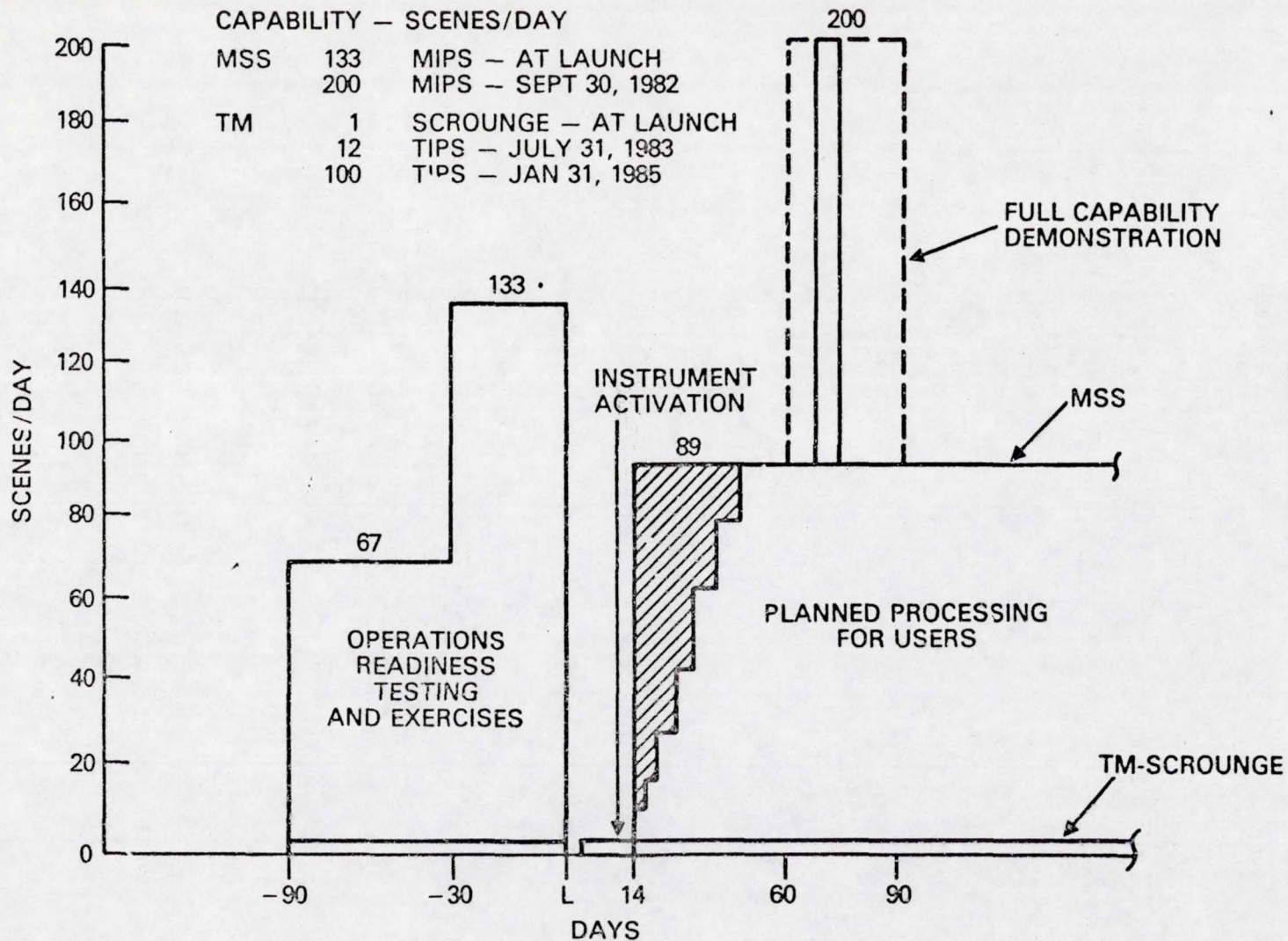
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TM IMAGE PROCESSING SYSTEM INTERFACES



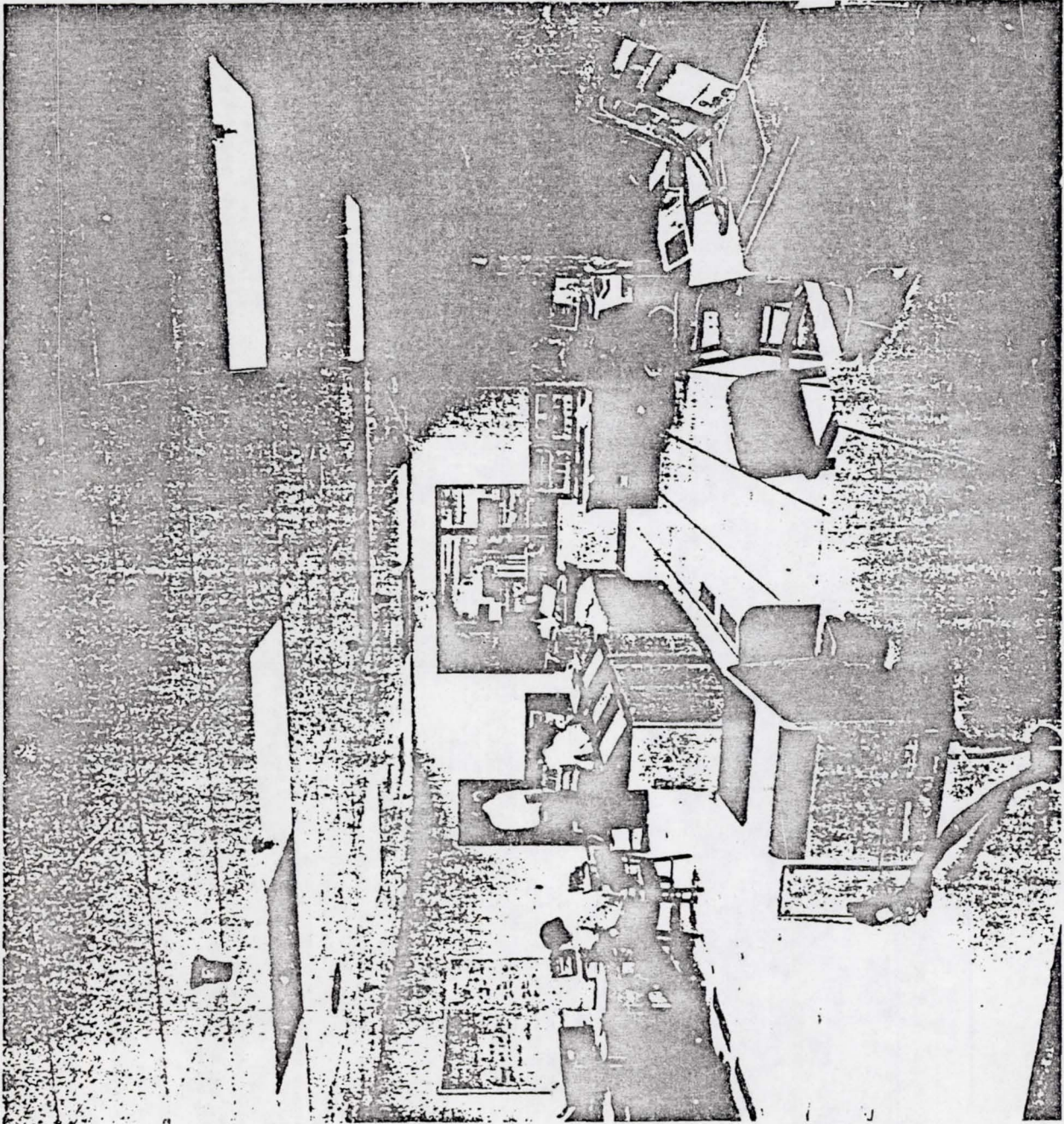
Processing



CONTROL AND SIMULATION FACILITY MAJOR FUNCTIONS

- **PLAN AND SCHEDULE FLIGHT SEGMENT OPERATIONS**
- **SCHEDULE LINK SUPPORT WITH NETWORK CONTROL CENTER**
- **COMMAND FLIGHT SEGMENT**
- **ACQUIRE FLIGHT SEGMENT TELEMETRY**
- **MONITOR, EVALUATE, AND REPORT FLIGHT SEGMENT PERFORMANCE**
- **SIMULATE FLIGHT SEGMENT OPERATION**
- **REPROGRAM ON BOARD COMPUTER**
- **PERFORM SELF TEST**

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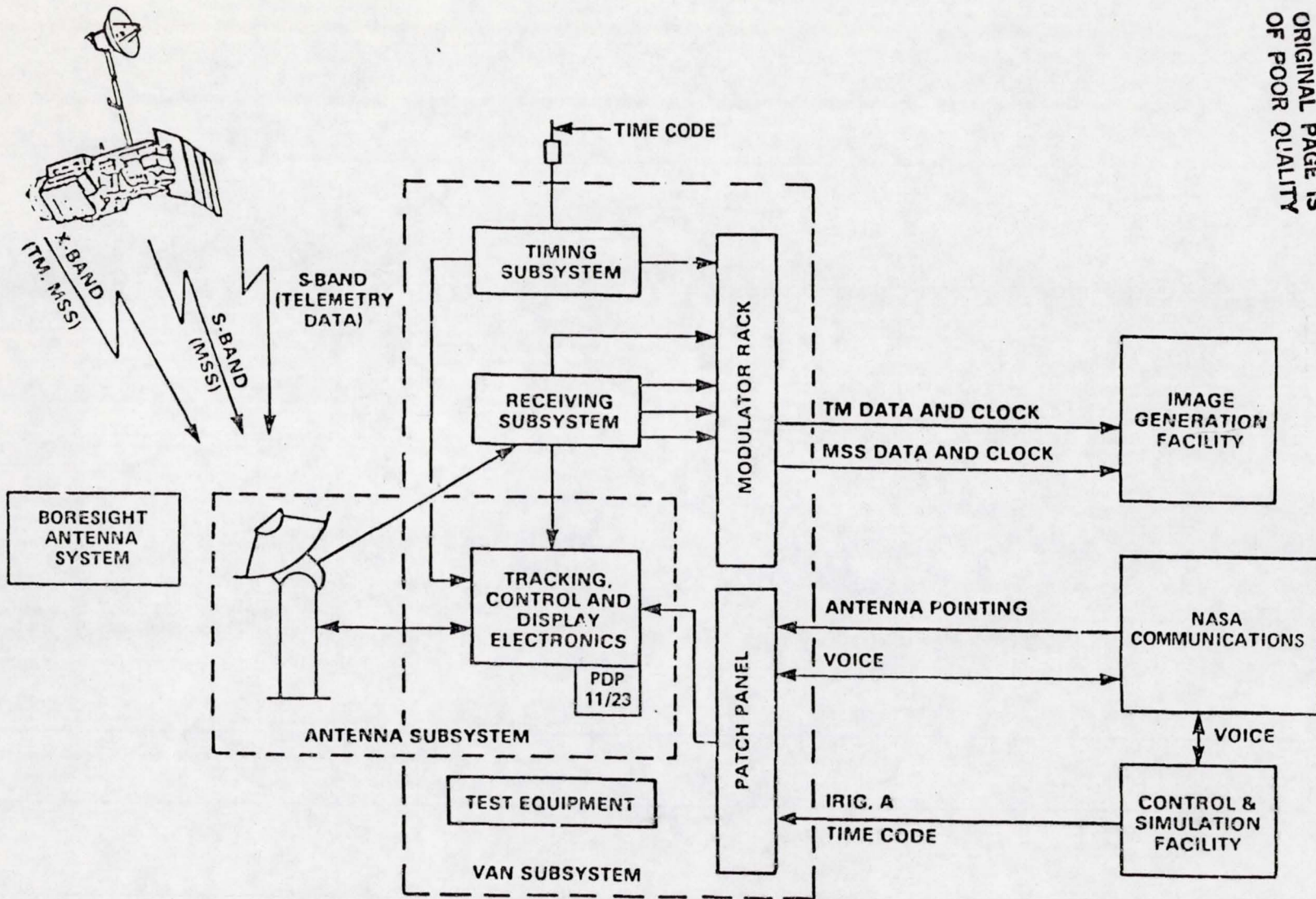
TRANSPORTABLE GROUND SYSTEM MAJOR FUNCTIONS

- **ACQUIRE TM/MSS VIDEO DATA**
 - **OPERATIONALLY IN PRE-TDRSS ERA**
 - **SUPPORT EVALUATION IN POST-TDRSS ERA**

- **PROVIDE TM/MSS VIDEO DATA TO THE IMAGE GENERATION FACILITY**

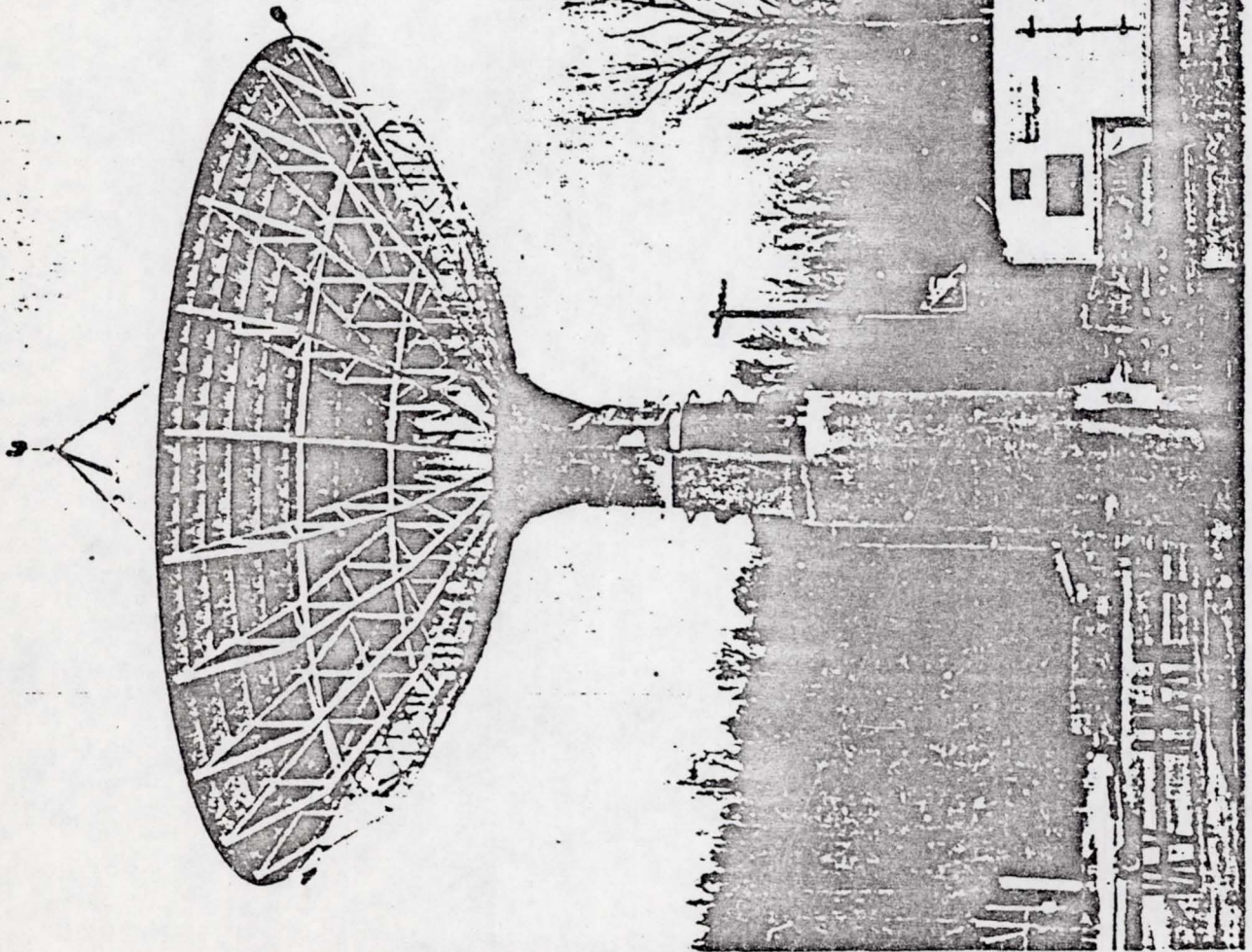
- **PROVIDE CAPABILITY TO RECEIVE NARROWBAND TELEMETRY**

TRANSPORTABLE GROUND STATION



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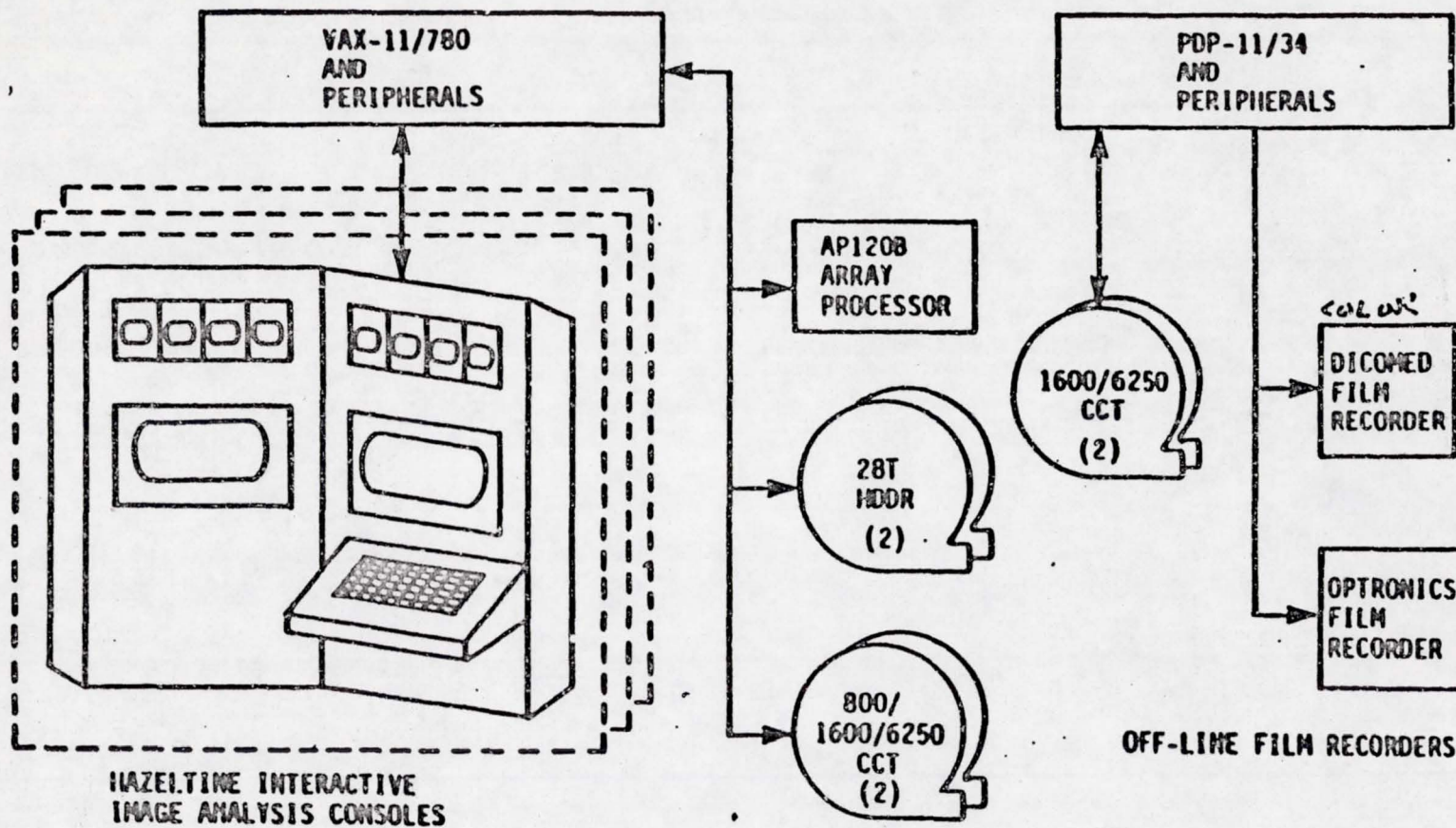
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LANDSAT ASSESSMENT SYSTEM MAJOR FUNCTIONS

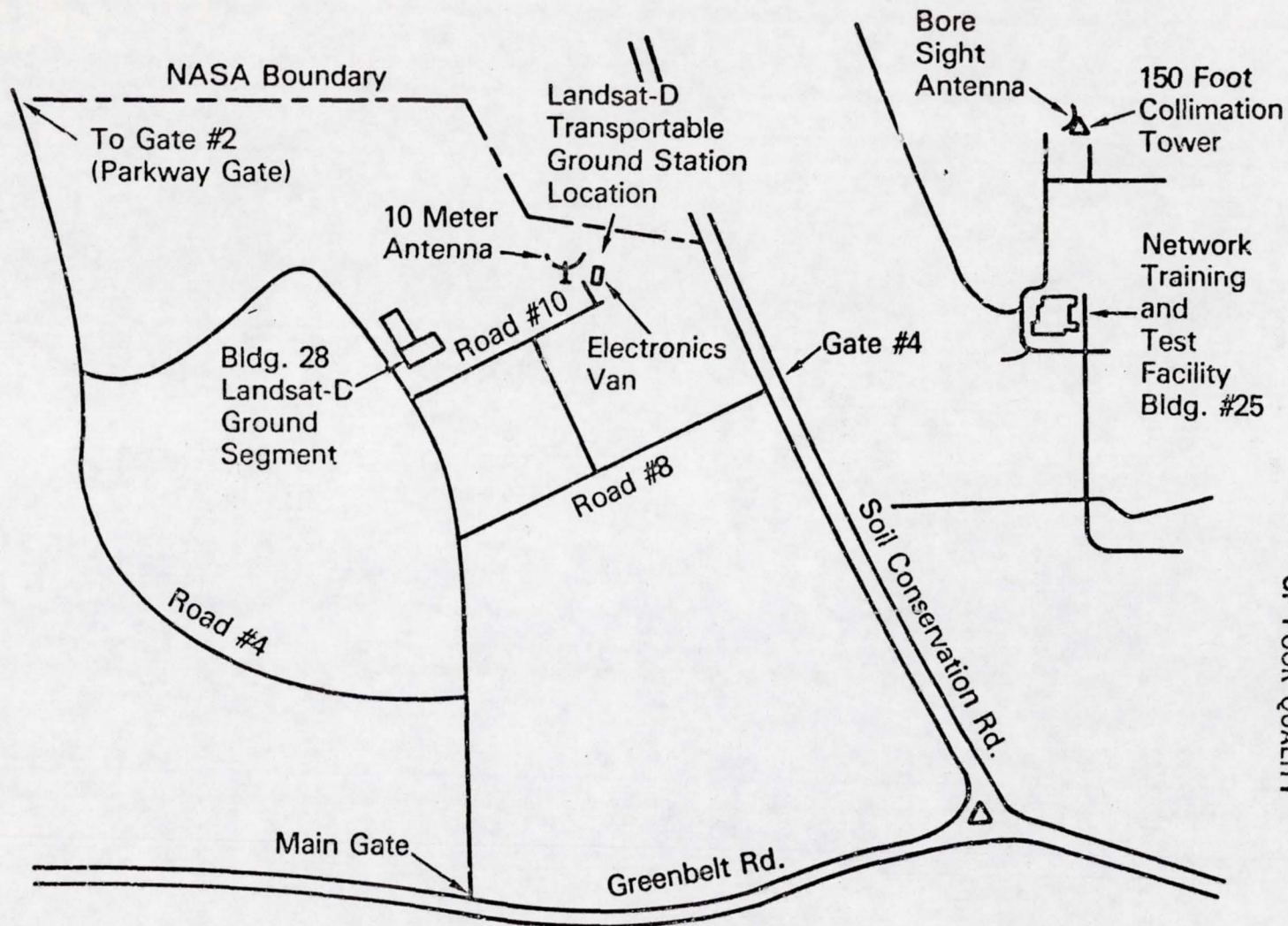
- PERFORM DATA QUALITY AND IMAGE SCIENCE TESTS OF TM AND MSS
- COMPARE TM AND MSS SENSORS
- COMPARE LANDSAT-D TO PREVIOUS LANDSAT MISSIONS
- PERFORM LIMITED APPLICATIONS INVESTIGATIONS IN SELECTED DISCIPLINES

LANDSAT ASSESSMENT SYSTEM
HARDWARE OVERVIEW



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Landsat-D Ground Segment Location



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Landsat-D Performance Characterization

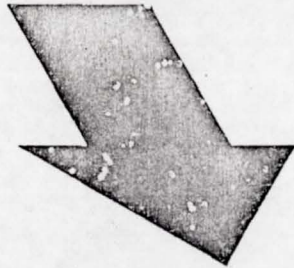
John Barker

Landsat-D Performance Characterization Activities

- Introduction
- Objectives
- Structure
- Engineering Verification
- Science Characterization
- Activities Schedule
- Investigations Workshop

Introduction

Landsat-D Performance Characterization is a Cooperative, Two-Part Engineering and Scientific Analysis Effort Designed to Foster the Effective Accomplishment of Overall Project Goals.



- Thematic Mapper (TM) Capability Assessment
- Multispectral Scanner (MSS) to TM Transition
- Operational System Feasibility Demonstration
- Continuity of MSS Imagery
- Continued Foreign Access

Objectives

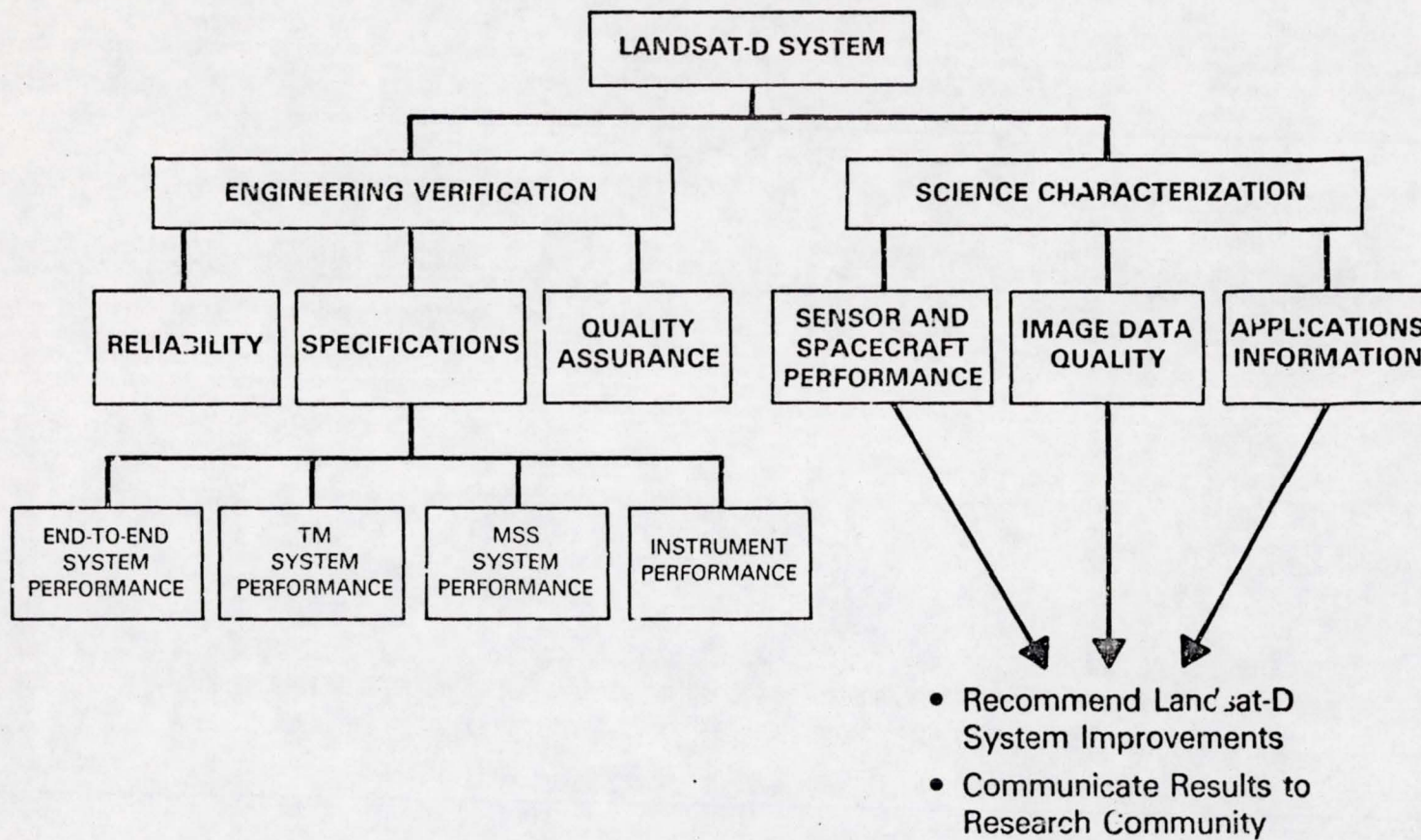
ENGINEERING:

- Verify Instrument, Data Processing Facility and Total System Performance to Specifications
- Establish Equipment and Operations Reliability
- Verify Product Quality Standards

SCIENCE:

- Characterize Accuracy and Precision of Sensor and Spacecraft Performance
- Characterize Accuracy and Precision of Image Data Quality
- Characterize Accuracy and Precision of Derived Information
- Recommend Landsat-D System Improvements
- Communicate Results to Research Community

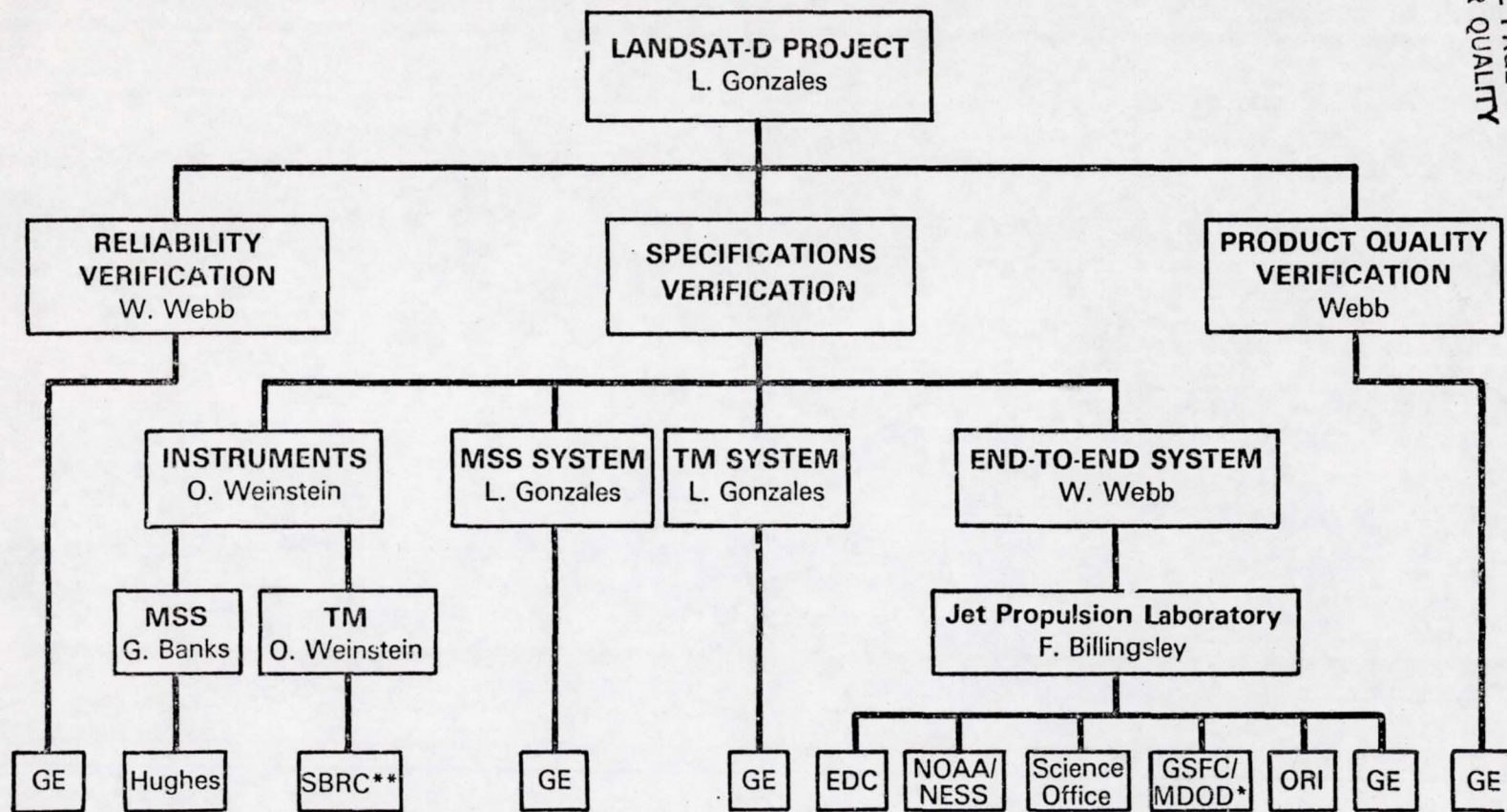
Structure



Engineering Verification

Engineering Verification Organization

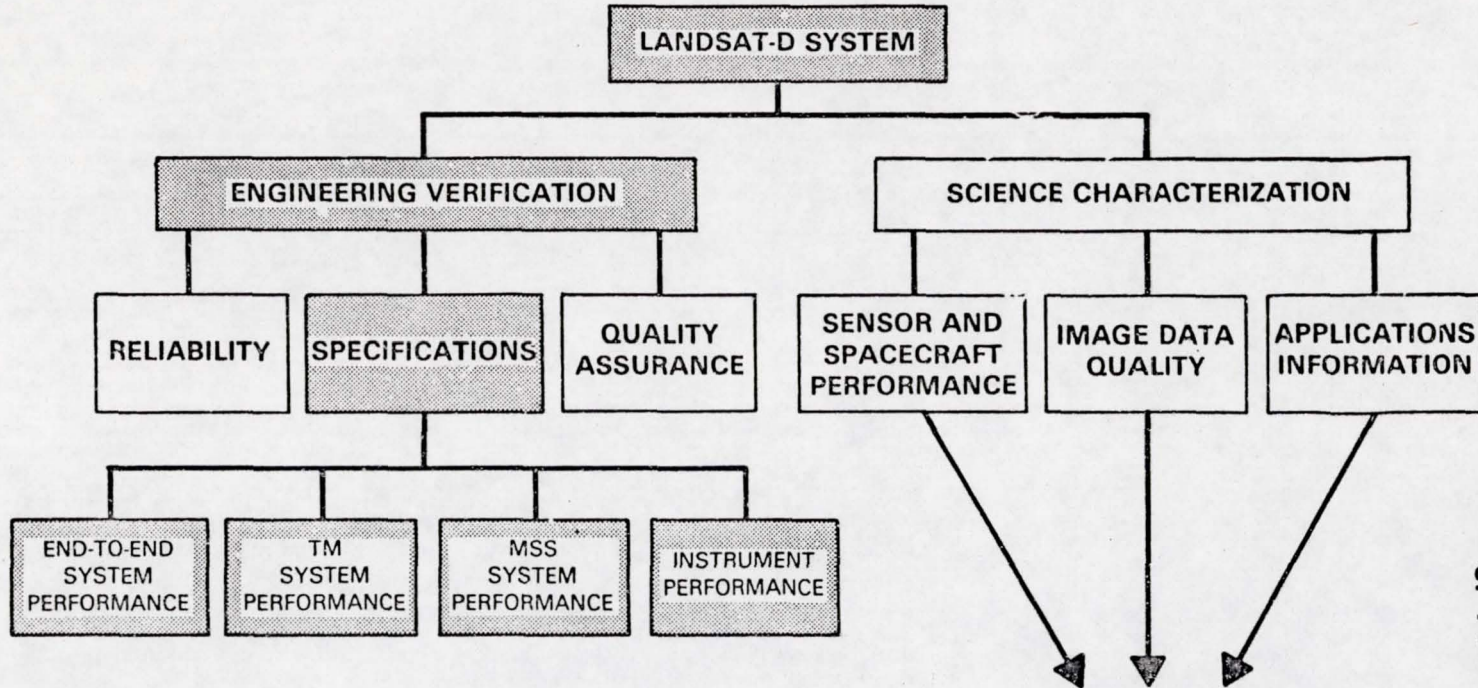
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* MDOD Mission and Data Operations Directorate

** SBRC Santa Barbara Research Center

Structure



- Recommend Landsat-D System Improvements
- Communicate Results to Research Community

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Instrument Performance Analysis

(SENSOR SYSTEM LEVEL TESTS)

RESPONSIBILITIES

MSS-Protoflight and Flight
(PF and F)

Hughes

TM (PF and F)

Santa Barbara Research Center
(SBRC)

REPORTS

- Technical Memos
- Pre-Ship Review
- Final Report

- Technical memos
- Pre-Ship Review
- Post-Launch Support

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MSS and TM System Performance Analysis

GE RESPONSIBILITIES

Pre-Launch

TM Radiometric Testing
and Data Reduction

TM Geometric Testing

Post-Launch

Radiometric Calibration
and Validation
Geometric Calibration
and Validation

REPORTS

- Technical Memos
- Pre-Ship Review
- Processing White Papers
and Data Reduction

- Technical Memos
- Post-Launch Support
- Processing Parameter
Update

End-to-End System Performance Analysis

RESPONSIBILITY

Fred Billingsley, JPL

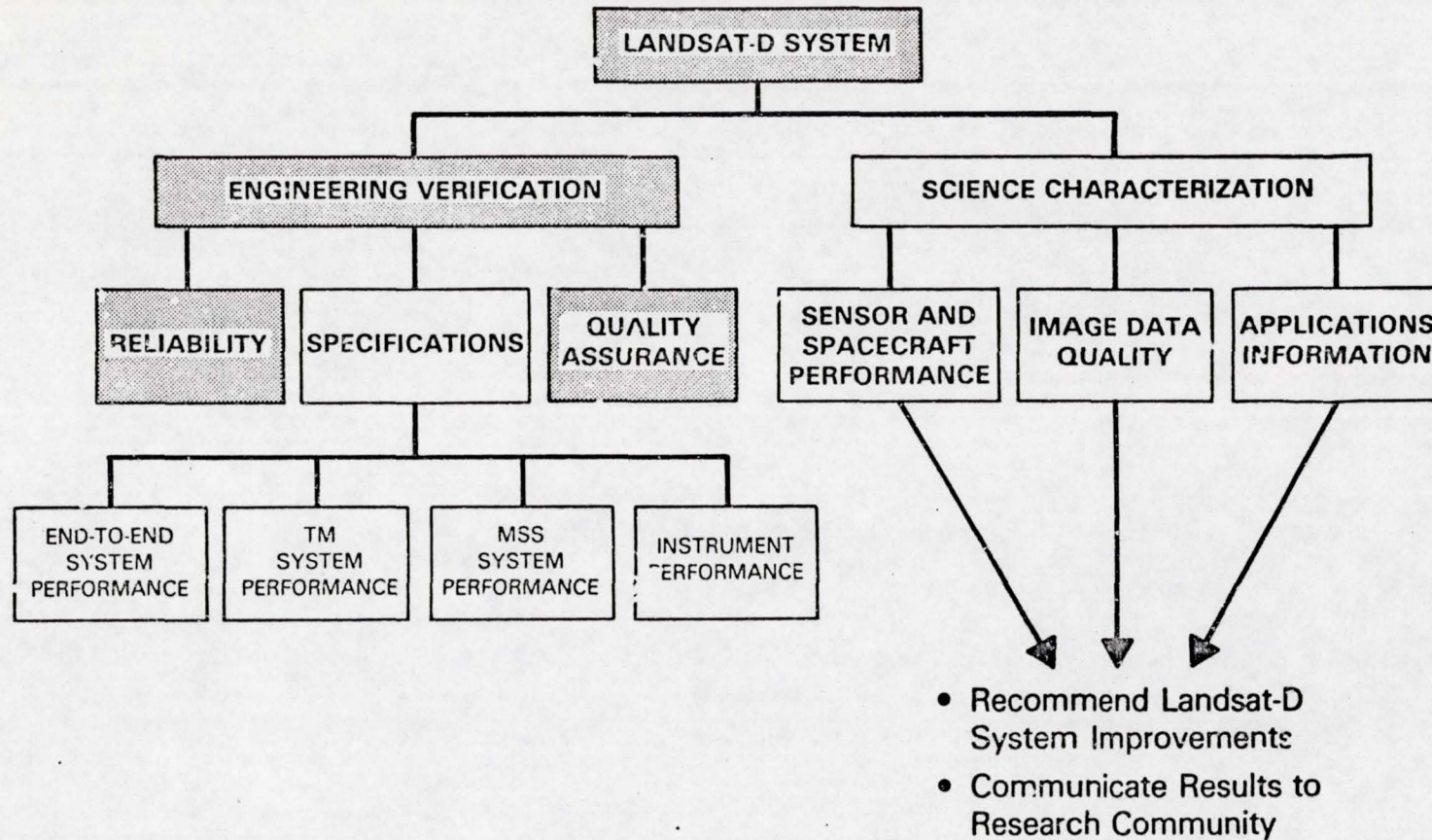
REPORTS

Pre-Launch Publication of
Landsat-D End-to-End
System Performance
Study

STUDY OBJECTIVES

- Determine to What Extent Intended System Performance is Possible
- Estimate Image Technical Performance to be Expected
- Determine if Adequate Ancillary Information is Present
- Trace Effects of System Functions and Operations Through the System
 - Determine End-to-End System Operability
 - Estimate Cumulative Errors

Structure



Reliability Verification

RESPONSIBILITY (ONGOING)

GSFC Mission Operations
Manager (Webb)
GE Engineering Support
Organization
GE Quality Assurance
Organization

REPORTS

Maintenance Plan
Configuration Management Plan
Equipment Service Reports (ESR)
Problem Defect Reports (PDR)
Management Reports

- Utilization
- Mean Time to Repair
- Inventory and Supplies
- Production Statistics
- Maintenance Histories

Quality Assurance Verification

RESPONSIBILITY (ONGOING)

GSFC Mission Operations
Manager (Webb)

GE Quality Assurance
Organization

- Product Evaluation
- Process Evaluation

REPORTS

Quality Assurance Reports

Processing Summary Reports

Automated PDR and ESR
Management Reports

System Audit Reports

Special Management Reports
(Trend Analyses & Current Quality
Problem Daily Report)

Quality Verification Supporting Information

INFORMATION	SOURCE
• "Approaches to Satisfying Landsat Data User Concerns"	• The Operational Landsat-D Quality Assurance/ Performance Evaluation Plan Review Meeting, February 1982 (Final Report)
• Performance Evaluation Product Generation	• Landsat-D Mission Operations Review (MOR), April 6-7, 1982
• Operational Quality Assurance	• MOR, April 6-7, 1982

TABLE 1: APPROACHES TO SATISFYING LANDSAT DATA USER CONCERNS

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USER CONCERNS	REQUIREMENTS	APPROACH TO SATISFYING REQUIREMENTS
<p>TIMELINESS OF DATA:</p> <p>Throughput</p> <p>Turnaround Time</p>	<ul style="list-style-type: none"> • 200 New MSS Scenes/Day • 48 Hours from Receipt of Data 	<ul style="list-style-type: none"> • High Level of Back Up in Design • Most Equipment Off-the-Shelf, Proven Performance • Availability/Reliability-Maintenance, Spares Philosophy • General Purpose System Simulator (GPSS) Model: <ul style="list-style-type: none"> - Timeliness, Utilization, - Failure Mode Effects Analysis, - Optimization of Operating Scenarios • Automatic Monitoring, Reporting of Performance
<p>COMPLETENESS OF DATA:</p> <p>Treatment of Data Dropouts, Errors</p>	<ul style="list-style-type: none"> • Replace Lost Data with Last Good Data • Detect and Correct Bad Time Codes • Report All Substitutions on Digital Products 	<ul style="list-style-type: none"> • Request Retransmission of Bad Data Transfers from White Sands • Provision to Skip Bad Data on Scene Basis • Repetition of Last Good Video Data for Lost Video Line • Insertion of Flywheel Times for Bad Time Code • Automatic Reporting of Substitute Video Data in Line Quality Maps
<p>RADIO-METRIC QUALITY OF DATA:</p> <p>Absolute and Relative</p>	<ul style="list-style-type: none"> • ± 1 Quantum Level over Entire Detector Range (6 Bit Data) within each Band 	<ul style="list-style-type: none"> • Calibration on Wedge for Absolute and Band-to-Band Fidelity • Nominal Calibration Wedge for Back-Up • Scene Content for Detector-to-

TABLE 1: APPROACHES TO SATISFYING LANDSAT DATA USER CONCERNS (Cont'd)

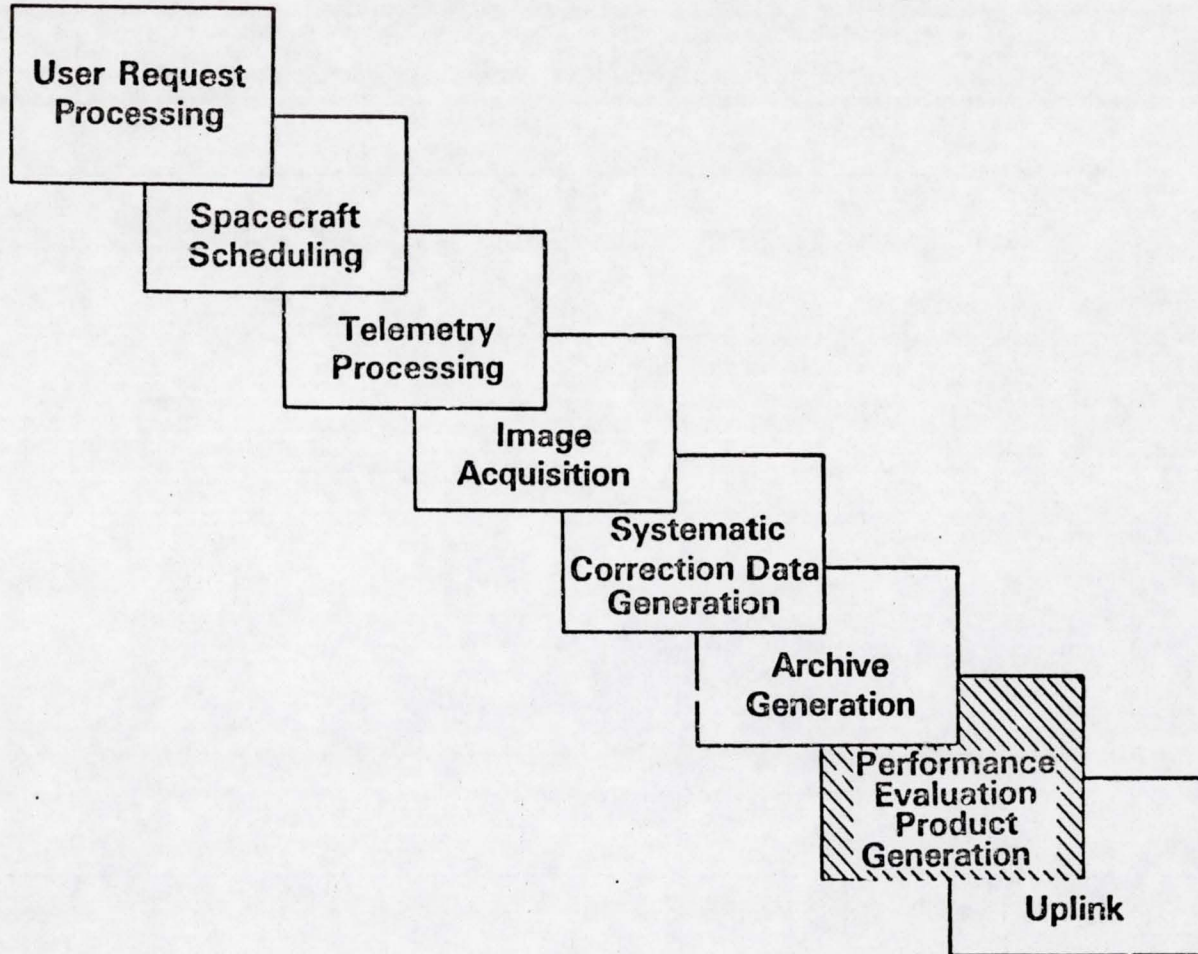
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USER CONCERNS	REQUIREMENTS	APPROACH TO SATISFYING REQUIREMENTS
<p>RADIOMETRIC QUALITY OF DATA (Cont'd):</p>		<p>Detector Ranging (Destriping)</p> <ul style="list-style-type: none"> ● Techniques Demonstrated on LS-2 Data: <ul style="list-style-type: none"> - Meet ± 1 Quantum Level Detector Range - Means and Variances of Radiance Population Preserved for each Sweep ● Automated Assessment of Performance Part of Processing System
<p>GEOMETRIC QUALITY OF DATA:</p> <p>Absolute and Temporal</p>	<ul style="list-style-type: none"> ● 0.3 Pixel Temporal Registration Accuracy (90%) ● 0.5 Pixel Geodetic Accuracy (90%) 	<ul style="list-style-type: none"> ● Systematic Correction Data (SCD): <ul style="list-style-type: none"> - Models of Spacecraft/Sensor/Earth System - Ephemeris, Attitude Measurements - Residuals Include Measurement Errors, Unmeasured Attitude Components (All Frequencies), Scanner Alignment, Scan Repeatability ● Geodetic or Temporal Correction Data (GCD): <ul style="list-style-type: none"> - From Automatic Control Point/Control Point Neighborhood (CPN) Registration (After Systematic Correction of CPN) - Techniques for Detecting False Correlations - Filtering of Systematic Control Point (CP) Location Errors to Update SCD to GCD
<p>AVAILABILITY OF DATA QUALITY MEASURES:</p>	<ul style="list-style-type: none"> ● Daily and Continuing Readiness for Operations (Quality Assessment) 	<ul style="list-style-type: none"> ● Quality Measures for All Phases of Processing

TABLE 1: APPROACHES TO SATISFYING LANDSAT DATA USER CONCERNS (Cont'd)

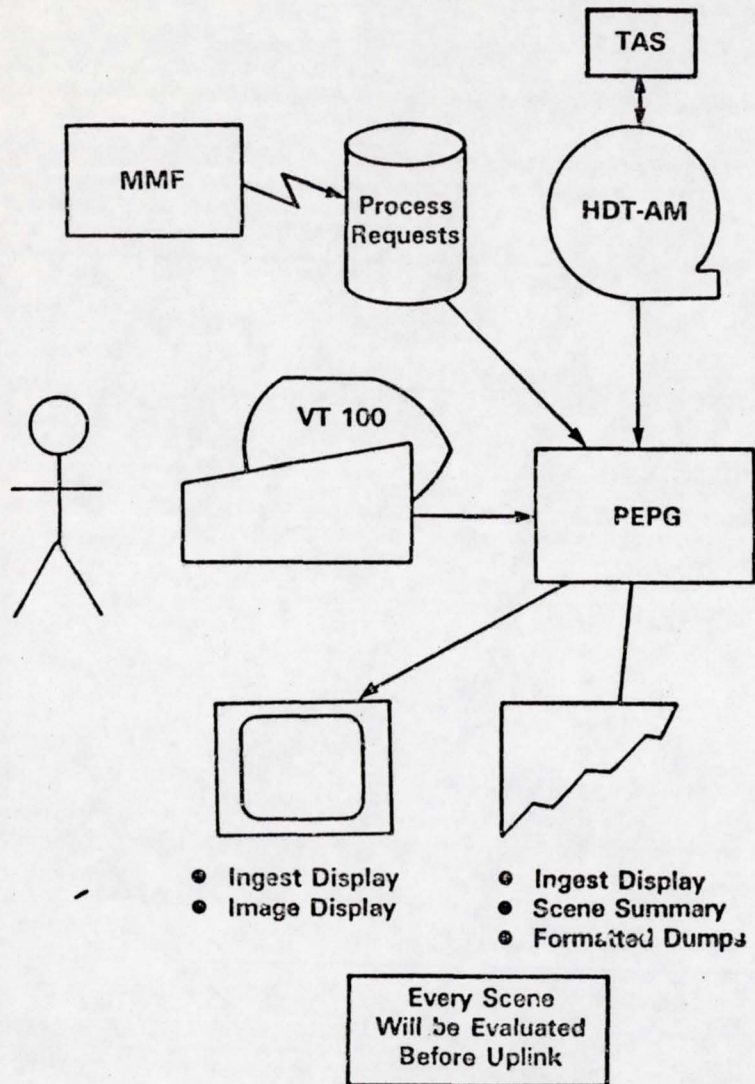
USER CONCERNS	REQUIREMENTS	APPROACH TO SATISFYING REQUIREMENTS
AVAILABILITY OF DATA QUALITY MEASURES (Cont'd):	<ul style="list-style-type: none"> ● Assurance Quality of Products ● Evaluation and Enhancement of Performance 	<ul style="list-style-type: none"> ● Some Quality Measures Evaluated Against Thresholds to Determine: <ul style="list-style-type: none"> - Rework - Rejection ● Long Term Evaluation and Enhancement of System

Standard MSS Processing



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PEPG—HDT-AM Evaluation



- Input Source/Process Request
- How/Manual Selection Via Menu for Automatic Processing
- Who/Computer Operator Using Standard Procedures
- Where/Any MIPS String

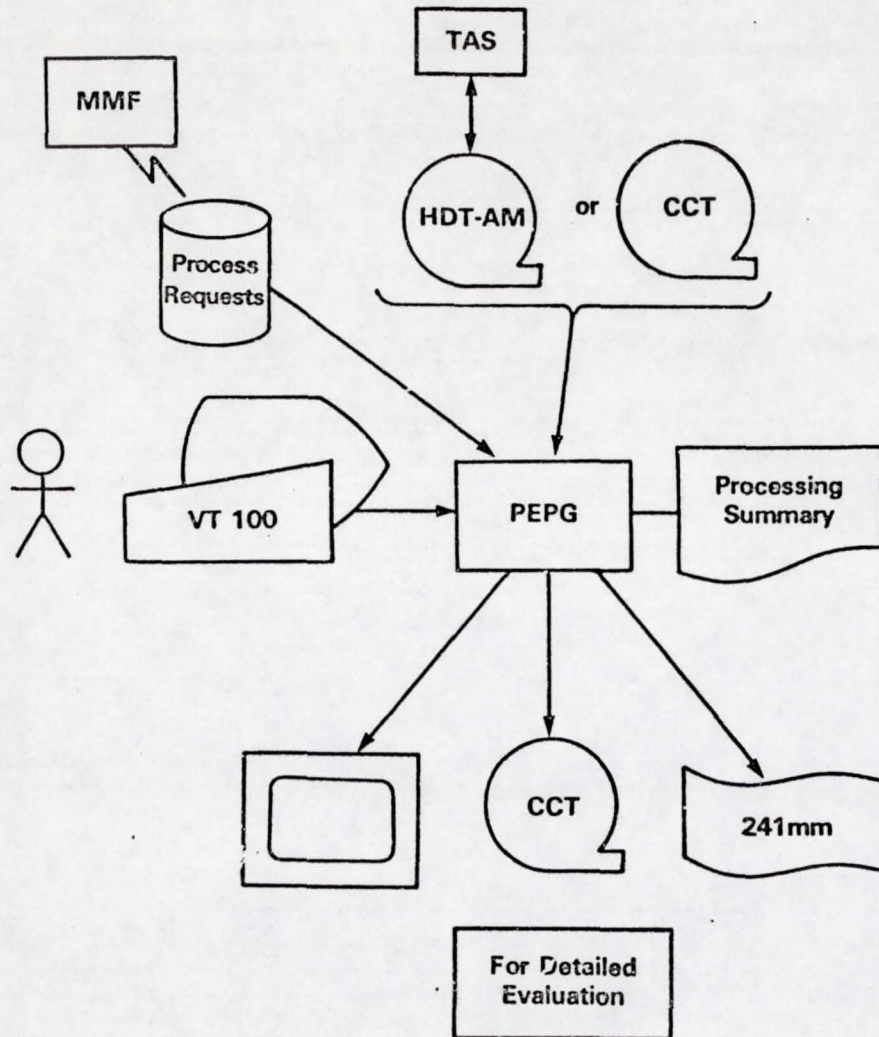
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HDT-AM Evaluation—Sequence of Events

- **Display Available Work**
- **Start Product Evaluation**
- **Operator Prompted for:**
 - **HDT-AM Mount**
 - **HDT-AM Dismount**
- **Ingest and Scene Summary Reports Generated Automatically**
- **Selected Scenes Stored on Disk for Evaluation by Quality Assurance**
 - **Formatted Dumps**
 - **Image Display**

PEPG Product Generation

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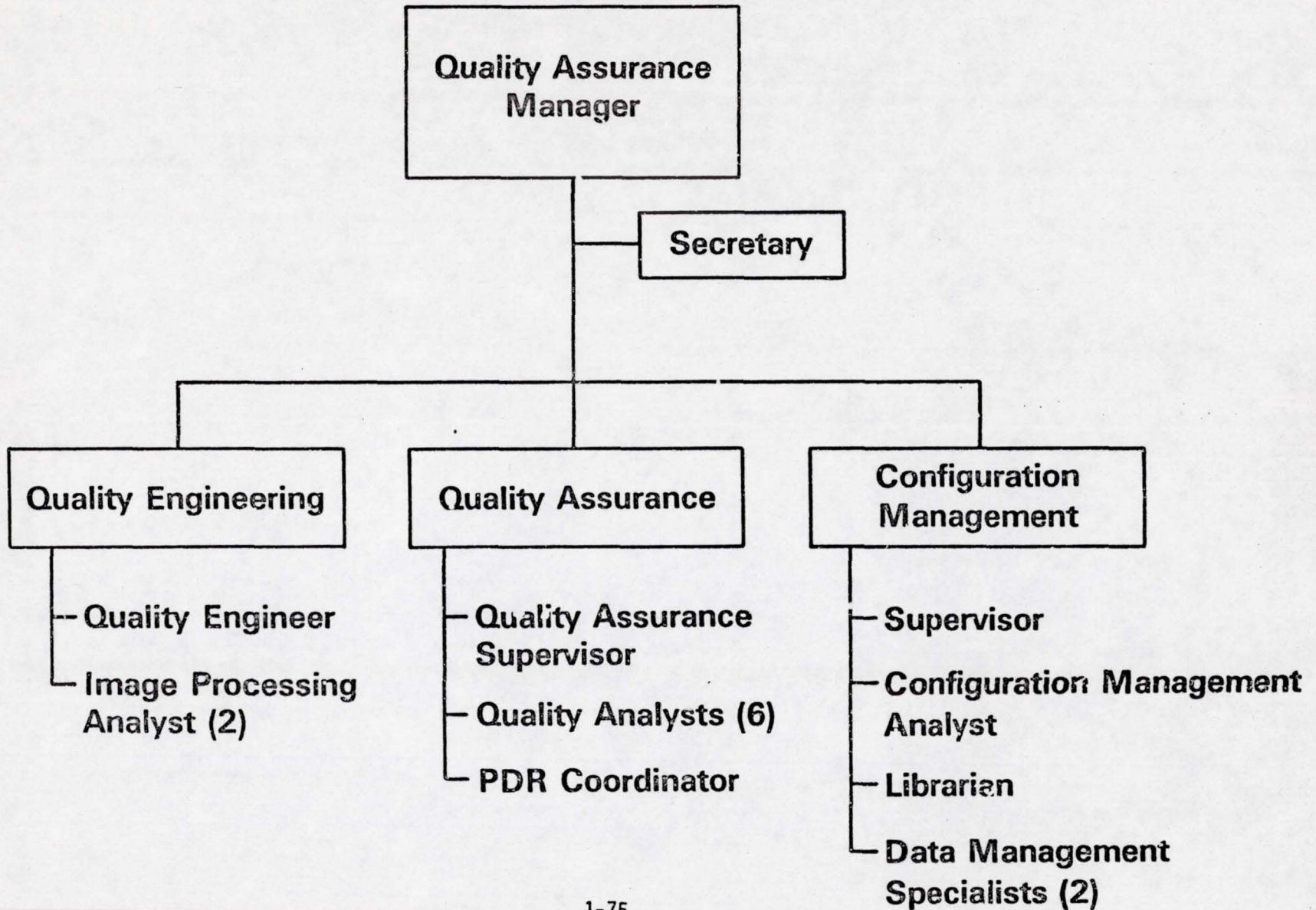
Input Source/Process Request

- How/Manual Selection Via Menu for Automatic Processing
- Who/Computer Operator Using Standard Procedures
- Where/MIPS for CCT (2 Scenes/Day)
TIPS for 241 mm (9 Scenes/Day)

Operational Quality Assurance

1-74

Quality Assurance Organization



Quality Assurance—Responsibilities

- Assure Performance — Measurement
— Evaluation
— Adjustment
— Enhancement

- Problem Management — Prevention
— Detection
— Investigation
— Solution
— Reporting

Quality Assurance Implementation

- **Quality Assurance Concepts**
- **Product Evaluation**
- **Process Evaluation**

Quality Assurance Concepts

- **Quality Assurance Features Designed Into System**
- **System is Fault Tolerant—Thruputs All Processable Data**
- **Fault Detection Built in, Limits Initially Set High**
- **System Captures Quality Indicators**
 - **Stored in MMF Data Base**
 - **Available in Many Computer Reports**
- **Quality Screening Responsibility Shared With Other Operators**
- **Quality Personnel Allocated for Problem Identification and Solution**
 - **Supported by Automated PDR/ESR System**
- **High Visibility to Management of Problem and Quality Reports**

System is Fault Tolerant

DRRTS

ECC's—Count Limit Checked ≤ 10 Uncorrectable (MSS) ≤ 1000 Correctable (TM)

— If Exceeded, Alarms for Operator; Summary in QA Report

Major Frame Sync Loss—If > 10 Consecutive, Automatically Breaks Interval

Bad Time Code—Identified in Directory

—Operator Instructed Via SOP to Re-Dub Good Time Code Data

Recording Quality From TGS—Displayed in Moving Window Display (Read After Write)

—Operator Response

- Notify TGS if Transmission Bad
- Switch Recorders if Recorder Problem

MMF

Quality Checks ECC's and Sync Loss Against Limits—Limits Initially Same as DRRTS/MIPS

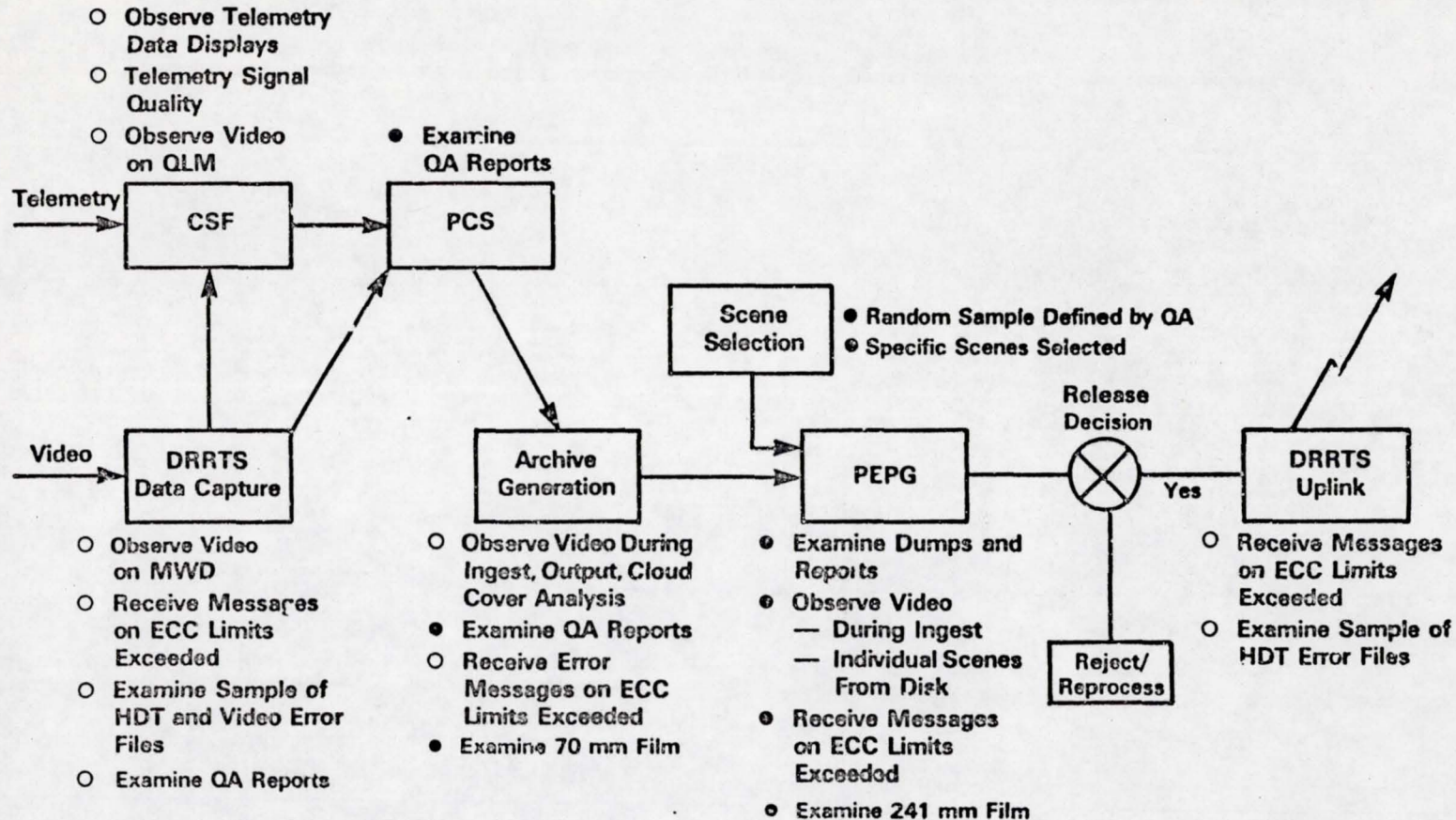
MIPS

ECC's—Same Alarms as DRRTS

Time Code—Substitutes if Can't Read

Sweep Substitution—Limit Checks—if Sync Loss for 10 Consecutive Major Frames, Declares Partial and Continues to Next Scene

QA Scenerio for Normal Processing (HDT-AM Generation)



QLM: Quick Look Monitor
MWD: Moving Window Display
ECC: Error Correcting Code

○ — Performed by Operators
● — Performed by QA

Product Evaluation

Assess Image Quality

- Real Time by Quality Analysts Using Visual and Data Evaluation Techniques
- Real Time by CSF/DRRTS Operators Using Moving Window Display, Quick Look Monitor, and Evaluators Consoles
- Off Line by Image Processing Analysts Using Visual and Data Evaluation Techniques

Authorize Uplinking of Acceptable Products

- By Quality Analysts Following PEPG Process
- By Image Processing Analysts Following Detailed Evaluation of Rejected/Reprocessed Data

Establish Accept/Reject/Reprocess Criteria

- By Image Processing Analysts With Concurrence of Engineering Review Board
- Update Using Pre and Post Launch Experience

Investigate User Feedback

- By Image Processing Analysts With Response Thru Project Office

Image Quality Assessment—Visual Techniques

- Each Scene—Scrolling Video Display (PEPG)

- Evaluation Criteria

- Video Present
 - Anomalies in Video Data
 - Correlate Video Data With Operator Messages

- 1 Band/Scene to 70 mm Film Product

- Evaluation Criteria

- Presence of All Characteristics (E. G., Video, Annotation, Tick Marks, Scene ID)
 - Anomalies in Video Data (Striping, Line Starts, Sync Loss)
 - Correlation With QA Reports

- PEPG —Upon Request by Image Processing Analyst

- Detailed Evaluation Using Comtal Display and 241 mm Film
 - Typically Used for—
 - More Thorough Evaluation of Apparent Problems Observed During Process
 - Investigation of PDR's
 - Precise Measurements to Support Performance Analysis

Image Quality Assessment—Data Analysis Techniques

- **Uses "Quality Indicators" Designed Into System**

- **Data Available From--**

Various Processing Reports

Tape Annotation Records

QA Reports

MMF Quality Files

Quality Indicators Used Real Time

Limit Checks in Software

Correlate to Video Display During PEPG

Accept/Reject/Reprocess Criteria Established in SOP's

Annotate Products for Users

Used Off Line

To Aid in Problem Investigation

To Support Performance Trend Analysis

To Support Adjustments in Criteria—Accept/Reject/Reprocess

To Support Changes in S/W Limit Checks

Typical Quality Indicators

DRRTS — Image Quality Data File

Location DECNET Header Record (DRRTS → MMF)

Data — Major Frames Out of Sync
 Minor Frame Sync Loss
 Minor Frame Sync Bit Errors
 Bit Slips

MAG QA Report — By Scene

 Radiometric Quality — Detector Data
 Summary by Band

MAG Processing Summary Report — By Scene

 Band Quality Indicators — Derived From
 Minor Frame Sync Loss
 Major Frame Sync Loss
 Line Substitutions
 Missing Line Starts

Overall and Quality Code

- Located in Byte 146 of HDT Header Data
- Ranges From 0 - 9, A,B,C 0 — Acceptable
C — Best
- f (Geometric Correction Quality, Radiometric Correction Quality, Image Data Quality)
- Geometric Correction Quality

<u>Code</u>	<u>Parameters Modeled</u>
A	No Parameter Modeled
G	Along Track, Across Track (Uses Control Points for Translation Errors)
E	Along Track, Across Track, Yaw, Altitude
- Radiometric Correction Quality (RCA—Max. Difference Between Detector Means for Image)

<u>Code</u>	<u>Criteria</u>
E	$0 < \text{RCA} < 1.0$
G	$1.0 < \text{RCA} < 2.0$
A	$2.0 < \text{RCA}$
- Image Data Quality $\left(\text{DQI} = \frac{\text{MJ FSL} + \text{MiFSL}}{20} + \frac{\text{Unrecov. ECC}}{20} \right)$

<u>Code</u>	<u>Criteria</u>
E	$0 \leq \text{DQI} \leq 1.5$
G	$1.5 \leq \text{DQI} \leq 4.5$
A	$4.5 < \text{DQI}$

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Process Evaluation

Problem Investigation

- PDR Investigations by Quality Analysts and Image Processing Analysts
- PDR Processing and Management Reports by PDR Coordinator
- Problem Trend Analysis By Image Processing Analysts Using PDR's and ESR's and Data Base

Process Quality Assessment

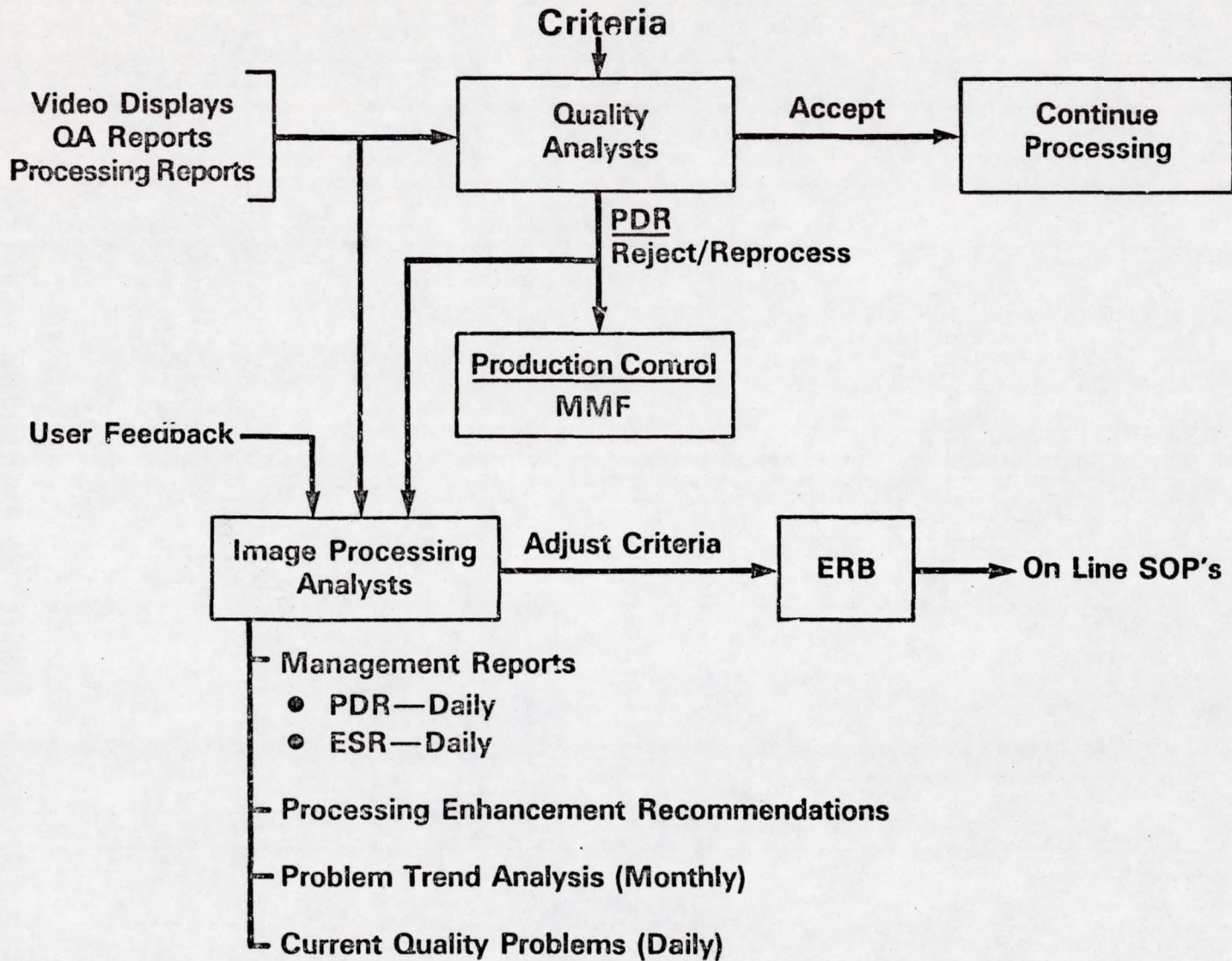
- Processing Success Evaluation by Image Processing Analyst Using Processing and QA Reports
- Operation Audits of All Functions by Quality Analysts
- Refinement of Use of Quality Indicators by Quality Analysts
- Processing Enhancement Recommendations
- Line Tests
 - Evaluate Results and Authorize Processing—Quality Analysts
 - Criteria Development and Evaluation—Image Processing Analysts (Approved by ERB)

Management Reporting

- Automated Management Reports for PDR's and ESR's
- Audit Reports
 - Immediate Reports to Responsible Manager
 - Corrective Action Reports Required
 - Management Report
- Special Management Reports
 - Problem Trend Analysis (Monthly)
 - Current Quality Problems (Daily)

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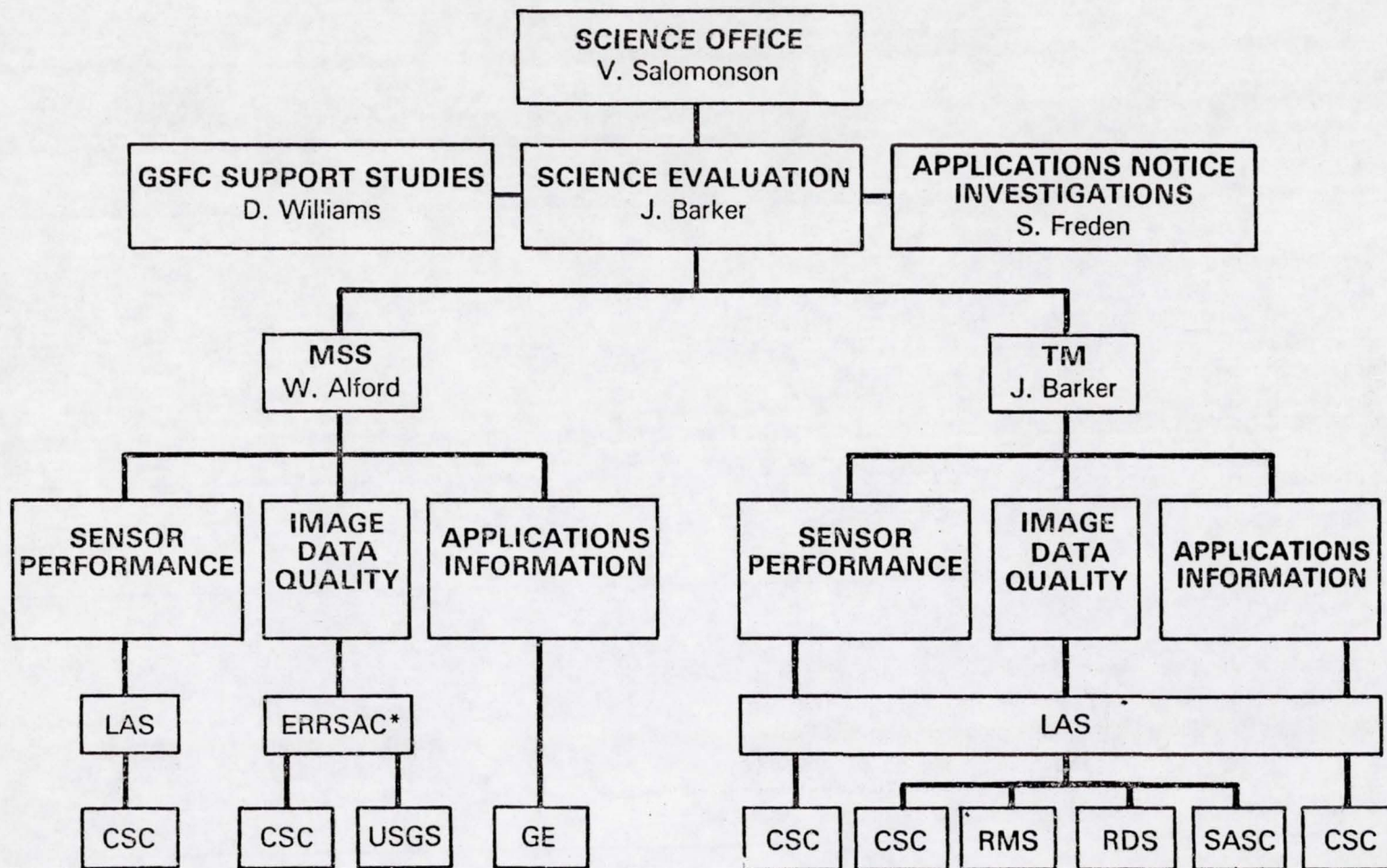
Accept/Reject/Reprocess Flow



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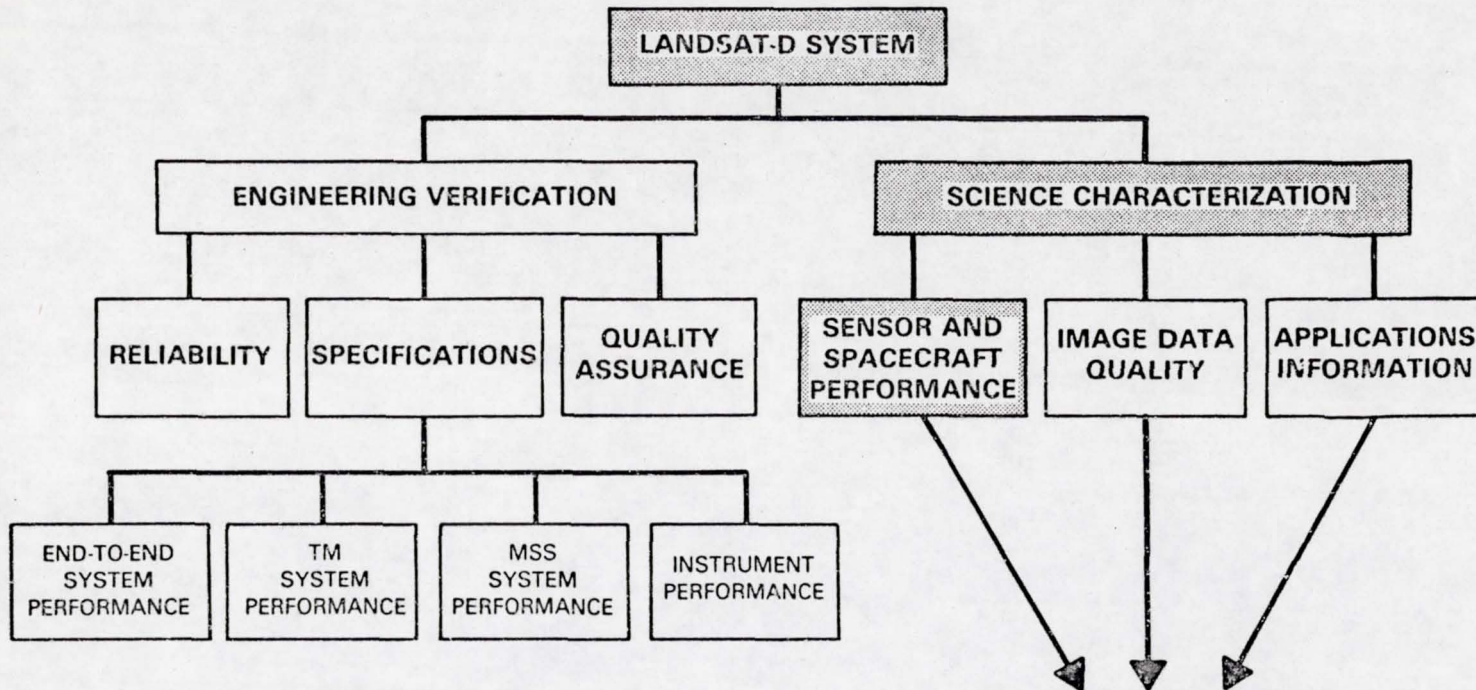
Science Characterization

Science Organization



*Eastern Regional Remote Sensing Application Center

Structure



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- Recommend Landsat-D System Improvements
- Communicate Results to Research Community

Sensor and Spacecraft Performance Characterization

AREAS OF INVESTIGATION

RADIOMETRY

A. Spectral Resolution

1. Filter
2. Detectors
3. System

B. Radiometric Resolution

1. Absolute Integrating Sphere Calibration (Dynamic Range
Linearity, S/N)
2. External Calibration (Precision)
3. Internal Calibration (Precision, S/N)
4. Flooding Lamp Calibration (Uniformity Over Scan)

Sensor and Spacecraft Performance Characterization

AREAS OF INVESTIGATION (CONT.)

GEOMETRY

A. Geometry of Pixel

1. Rise Time and Delay Time
2. Bright Target Recovery Time
3. MTF (IFOV) or Frequency Response Time
4. Bowtie Scan Angle Effect
5. Altitude Effects

B. Geometry of Image (Pixel Location)

1. Sensor Effects

A. Scan Profile for Reference Detector

- Along and Across Scan
- Forward and Reverse Scan

B. Detector Location Relative to Reference Detector

- Band-to-Band Registration
- Forward and Reverse Scan

C. Between Scan Alignment

- Reference Detector Forward and Reverse Offset
- Along and Across Scan

2. Ephemeris

A. Orbital Support Computing Division (OSCD)

B. Global Positioning System (GPS)

3. Attitude

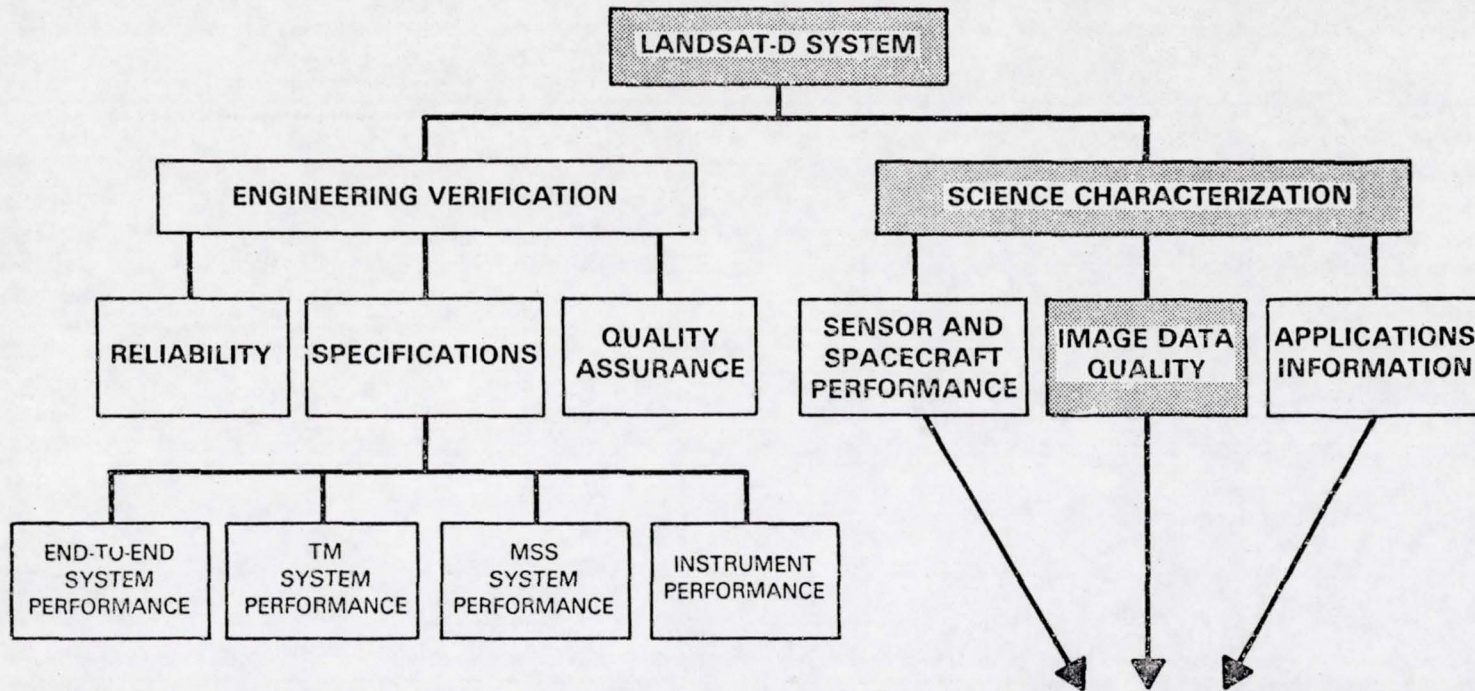
A. Angular Displacement Sensor (ADS)

B. Inertial Reference System (DRIRU)

C. Attitude Control System (ACS)

D. Alignment to Sensor (ADS, DRIRU and ACS)

Structure



- Recommend Landsat-D System Improvements
- Communicate Results to Research Community

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Image Data Quality Performance Characterization

AREAS OF INVESTIGATION

RADIOMETRY

A. Spectral Information

1. Detector Replacement Algorithms
2. Band Compression Algorithms

B. Radiometric Information

1. Internal Calibration Algorithms
 - A. Channel-to-Channel
 - B. Band-to-Band
2. Scene Histogram Calibration Algorithms
(Radiometric Destriping)
3. Absolute Scene Radiance Calibration Algorithms
 - A. Reflective Band
 - B. Thermal Band
4. Noise Correction Algorithms

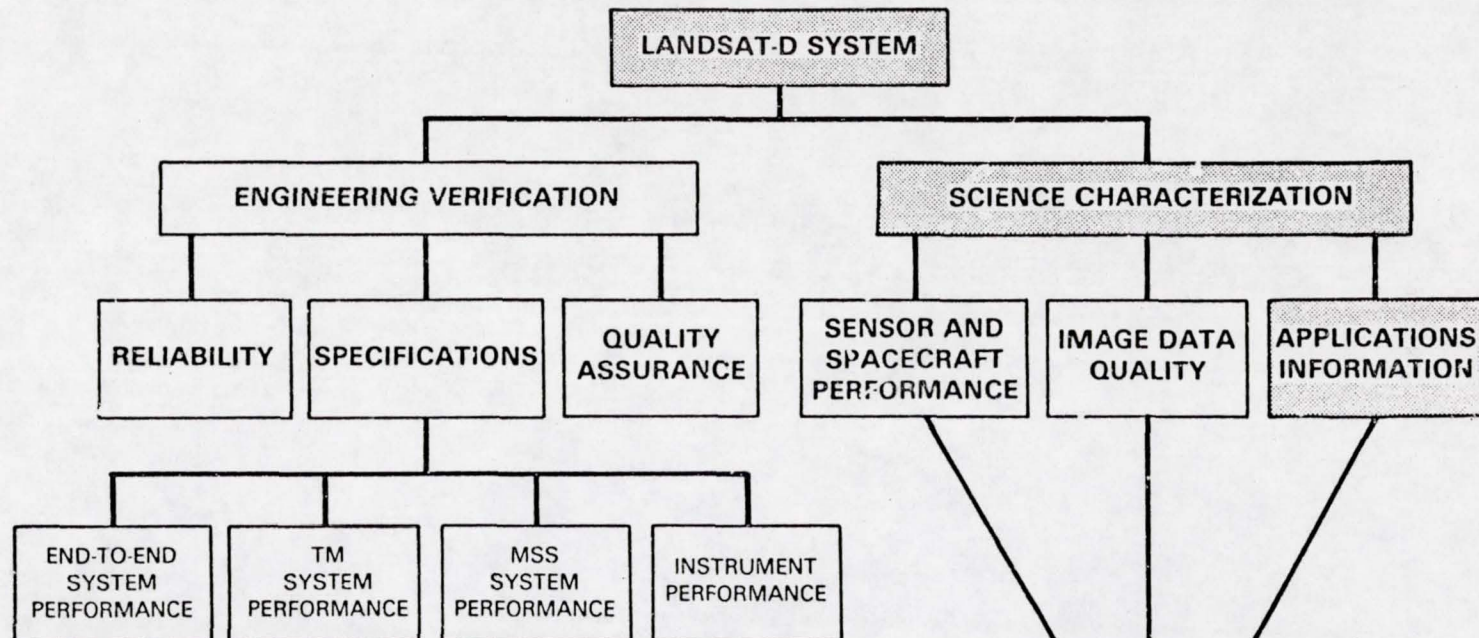
Image Data Quality Performance Characterization

AREAS OF INVESTIGATION (CONT.)

GEOMETRY

- A. Geometry of Pixel (Ground IFOV)
- B. Geometry of Image (Pixel Location)
 - 1. Systematic Correction
 - A. Scan Profile
 - B. Detector Location (Forward and Reverse Scan Alignment, Gap and Overlap)
 - C. Between Scan Alignment
 - D. Ephemeris
 - E. Attitude
 - 2. Geodetic Correction with Ground Control Points (GCPs)
 - A. Reference Library Build (Scene-to-Map Rectification)
 - B. Scene-to-Reference Scene Registration
 - 3. Resampling

Structure



- Recommend Landsat-D System Improvements
- Communicate Results to Research Community

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Applications Information – MSS and TM

Areas of Interest

RENEWABLE RESOURCES	NON-RENEWABLE RESOURCES	PLANNING/ENVIRONMENTAL MANAGEMENT
<p>Agriculture</p> <ul style="list-style-type: none"> Inventory Yield Condition Irrigation Episodal Event <p>Soils</p> <ul style="list-style-type: none"> Classification Erosion Moisture <p>Forests</p> <ul style="list-style-type: none"> Inventory Stand Evaluation Condition Episodal Event <p>Range</p> <ul style="list-style-type: none"> Vegetation Inventory Condition Episodal Event 	<p>Geology</p> <ul style="list-style-type: none"> Structure Landforms Lithology Thermal Anomalies Geobotanical Anomalies Topography (Stereo) Episodal Event <p>Image-Science</p> <ul style="list-style-type: none"> Pattern Recognition Information Extraction 	<p>Regional/Urban Land Use</p> <ul style="list-style-type: none"> Cover Classification Cover Change Environmental Impact <p>Coastal Zone</p> <ul style="list-style-type: none"> Monitoring <p>Hydrology</p> <ul style="list-style-type: none"> Drainage Patterns Inland Water Inventory Snow Pack Parameters Ice – Inland & Near Shore Water Quality – Inland & Near Shore Wetland/Estuaries Inventory Episodal Event <p>Wildlife Habitat</p> <ul style="list-style-type: none"> Inventory Evaluation <p>Oceans</p> <ul style="list-style-type: none"> Currents (Near Shore) Tides Bathymetric Charts Ocean Pollution (Near Shore)

Image Data Quality Performance Characterization - MSS

Geometry	Spectral Information	Detector Replacement Algorithms		Band Compression Algorithms		Channel to Channel Algorithms		Scene Histogram Calibration Algorithms (Radiometric, Destriping)		Absolute Scene Radiance Calibration Algorithms		Reflective Band Thermal Band		Noise Correction Algorithms		Ground IFOV		Systematic Correction		Geometric Correction with GCPs		Resampling			
		Internal Algorithms	Band to Band	Channel to Channel	Band to Band	Radiometric	Thermal Band	Reflective Band	Thermal Band	Noise Correction	Ground IFOV	Scan Profile	Detector Location	Between Scan Alignment	Ephemeris	Attitude	Reference Library Build	Scene to Reference Registration	Resampling						
		Name																							
		GSFC No	3	8	9	12	14	15	18	20	23	25	26	28	29	31	32	35	36	37	38	41	42	44	46
		Colewell	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Bernstein	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Arnita	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Mallia	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Bender	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Pride	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Slater	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Wrightley	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Mallia	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Zobrysi	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Erchson	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Hovis	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Laier	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Schott	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Kettler	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Walch	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Potter	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Guernsey	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Hill	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Everett	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		MacDonald	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN
		Anderson	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN	AN

Investigations Workshop Schedule

FIRST WORKSHOP TIMING

TODAY IS TOO EARLY:

- Still in AN Contract Negotiation
- Much More Landsat-D Information Still to Come

TODAY IS TOO LATE:

- Less Than Two Months to Launch
- Already Buried in Data !
- Already More Questions Than Answers !

Investigations Workshop Objectives

- Provide Pre-Requisite Information on:
 - MSS & TM Radiometry and Geometry
 - Data Acquisition, Processing and Availability
 - Nature and Direction of Investigations Program
- Create the Investigations Team
- Get Help:
 - Review of Draft Reports
 - Refinement of AN Areas of Investigation
 - Identification of AN Data and Information Requirements

Investigations Workshop Activities

PROVIDING INFORMATION

- Workshop Presentations
- Thursday Evening Tours
 - Five Offered (½ Hour Each)
 - Select up to Four During 2 Hours Allotted
- Supporting Documentation (Tape Formats, Draft Reports, etc.)
 - Order Via Form Provided
 - Limited Quantity Available Today
 - Remainder of Order Filled by Mail

CREATING TEAM

- Investigations Team Interaction Opportunities
- Principal Investigator Meetings with Science Representatives

Investigations Workshop Activities (Cont.)

ASKING FOR HELP, PLEASE:

- Identify Specific Additional Information Required, When Desired and When Needed (Via Workshop Return Form)
- Review Indicated Draft Reports (Via Workshop Return Form)
- Locate Individual Investigation Area Within Matrix (Via X on Matrix in Meeting with Science Representative)

Early Access TM Processing

John Lyon

Landsat-D Assessment System

SYSTEM OBJECTIVES

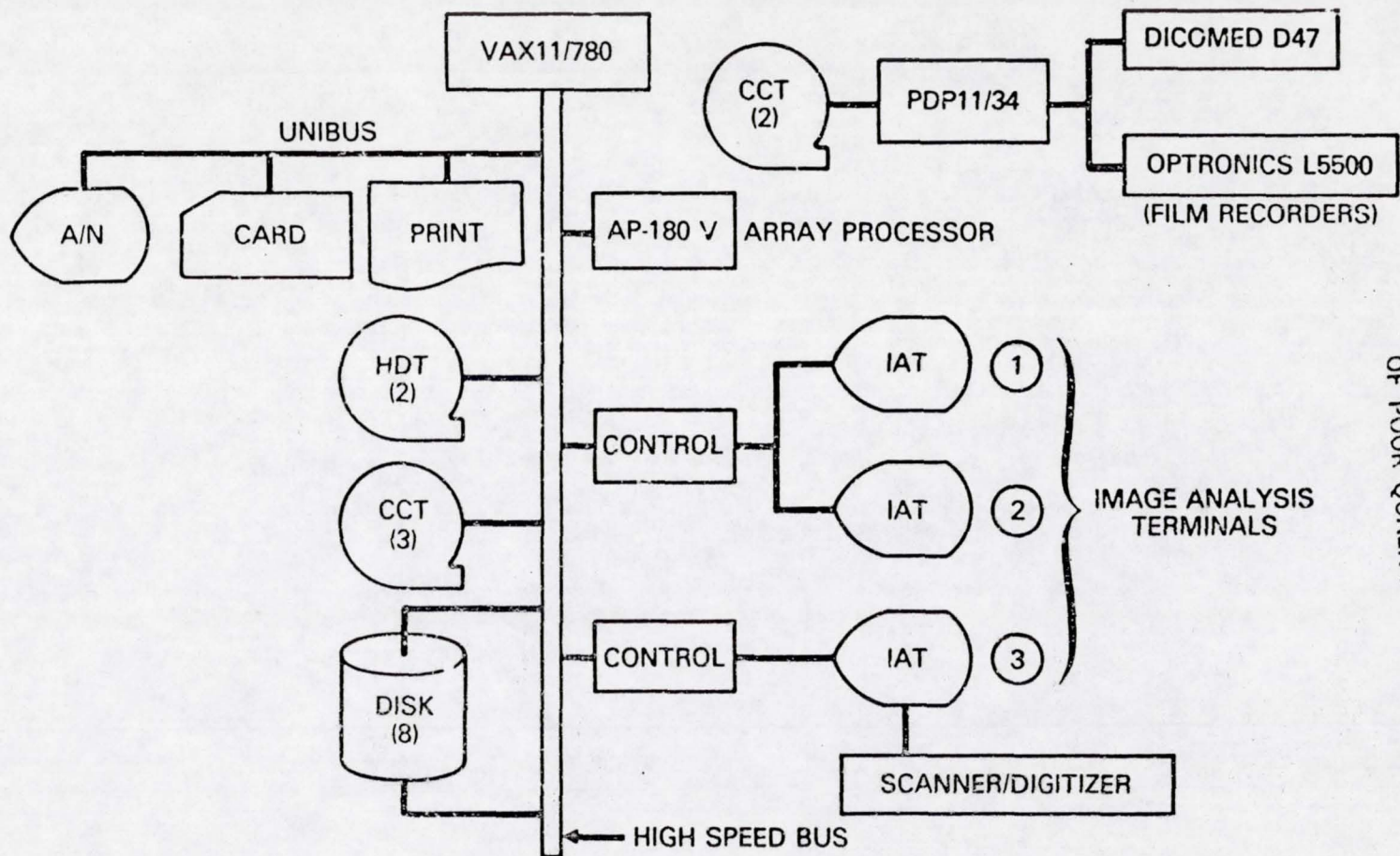
TM EARLY ACCESS PROGRAM (SCROUNGE)

- Provide, In Concert with the Applications Developmental Data System and Components of the Ground Segment, the Only TM Products Available in the First Year of Orbital Operations
 - One Scene/Day Using A Priori Corrections
 - Standard Products: P Film and Digital Data on 6250 BPI Tapes
 - Also Available: A and "B" Data Sets as Necessary

RESEARCH/ANALYSIS DATA SYSTEM ALLIED WITH GROUND SEGMENT

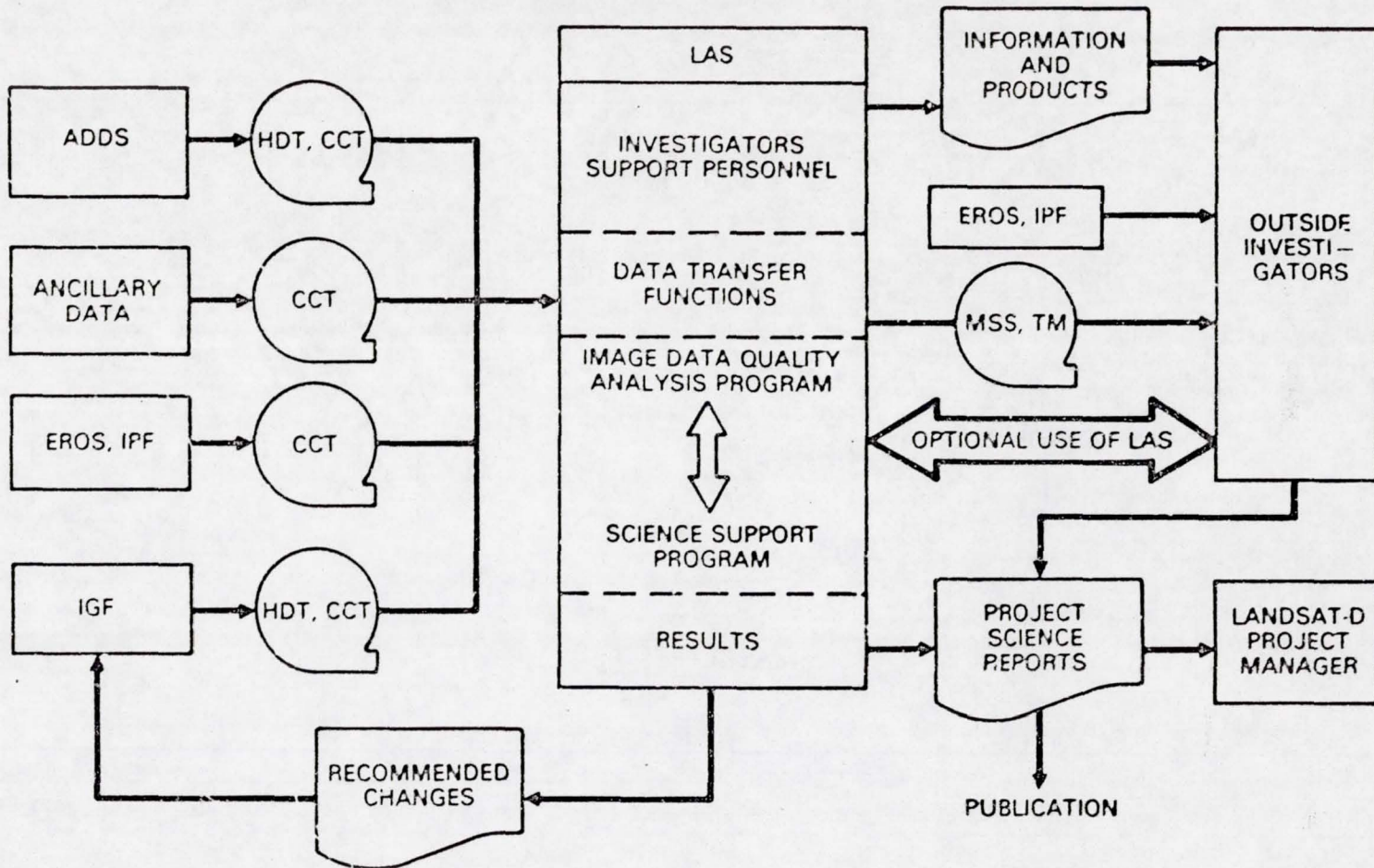
- Open Ended/Flexible
- To Accomodate Meaningful Data Quality and Interpretation Studies v. Instruments, Emphasizing TM
- Available for AN Support with Some Resource Contention with Scrounge

Overall Configuration



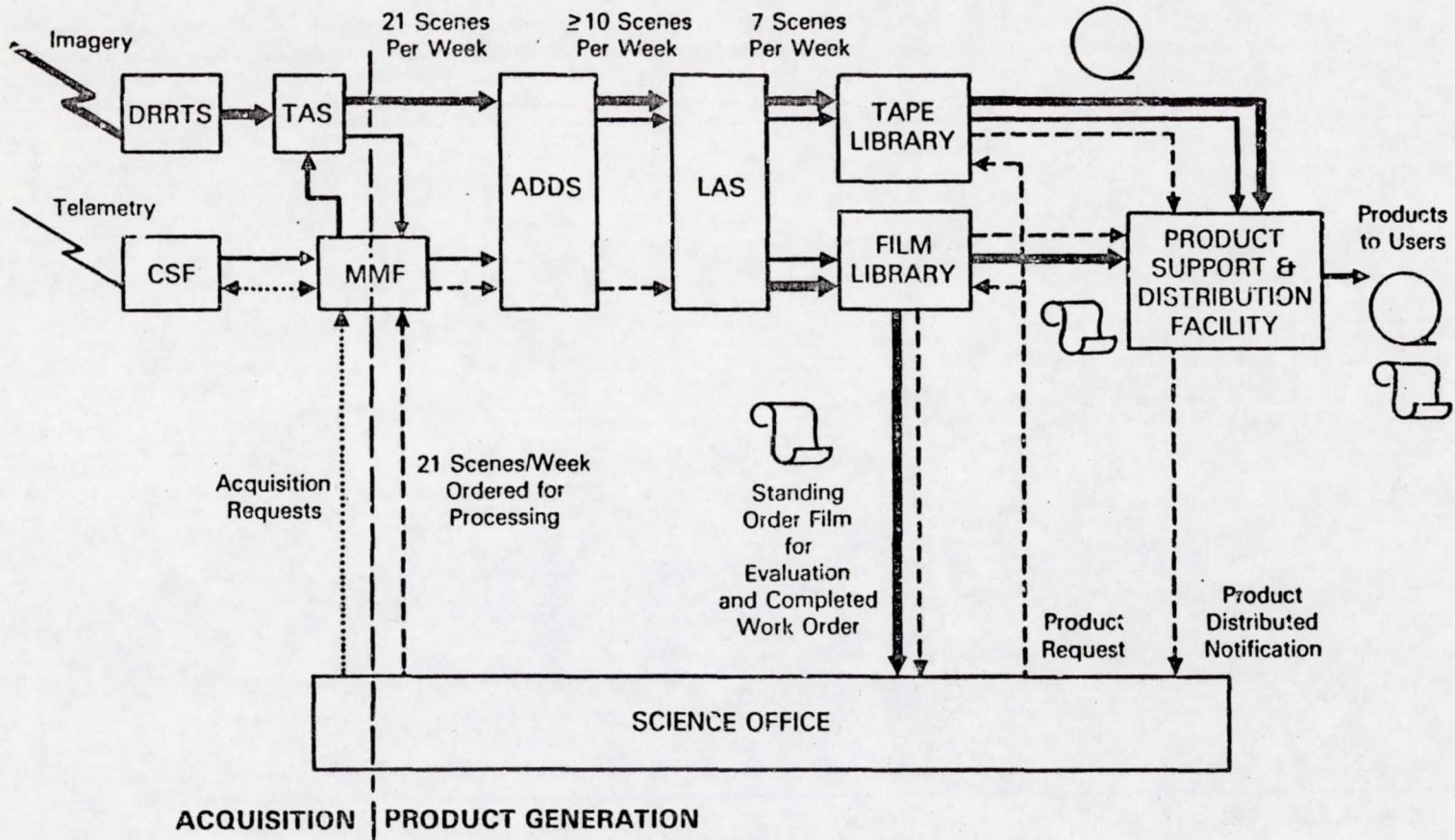
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Image Data Quality Analysis/ Science Support Program



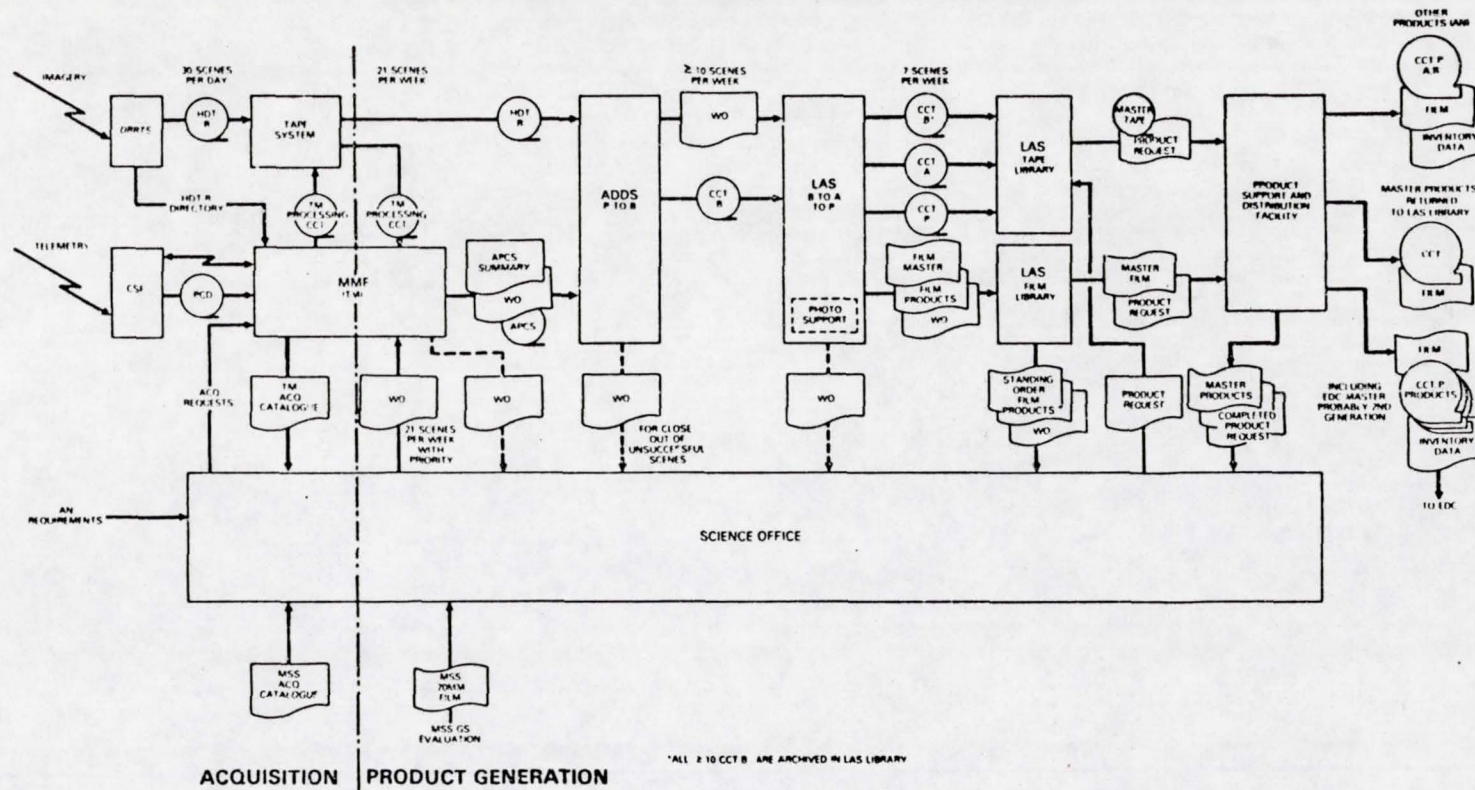
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End-to-End Scrounge Data Flow



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End to End Scrounge — LAS

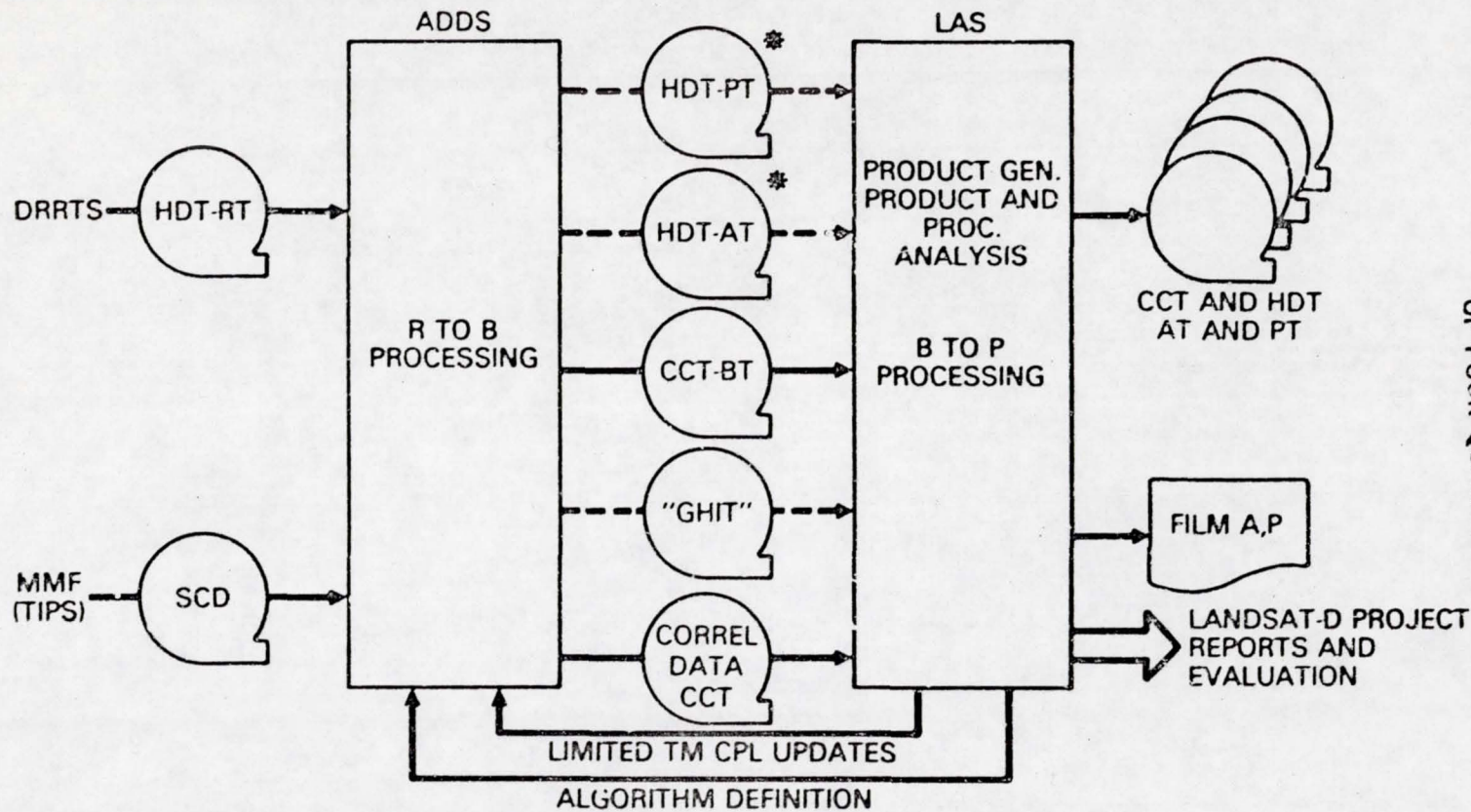


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LAS Functions

- Receive a Minimum of 10 and up to 21 TM Scenes Per Week in CCT-B Format
- Receive Corresponding Work Orders and Scene Priorities
- Apply Radiometric and Geometric Corrections to TM Data as Required to Produce CCT-A and P Products
- Produce TM P-Film Master and Associated Products for 7 Scenes Per Week
- Forward Standing Order Film Products and Updated Work Orders to Science Office
- Store Tape and Film Master in Respective Libraries
- Supply Film and Tape Masters to Products Support and Distribution Facility (According to Product Requests) for Preparation of Output Products
- Provide Science Office with Weekly Processing Summary Report

TM Early Access Program Functional Data Flow



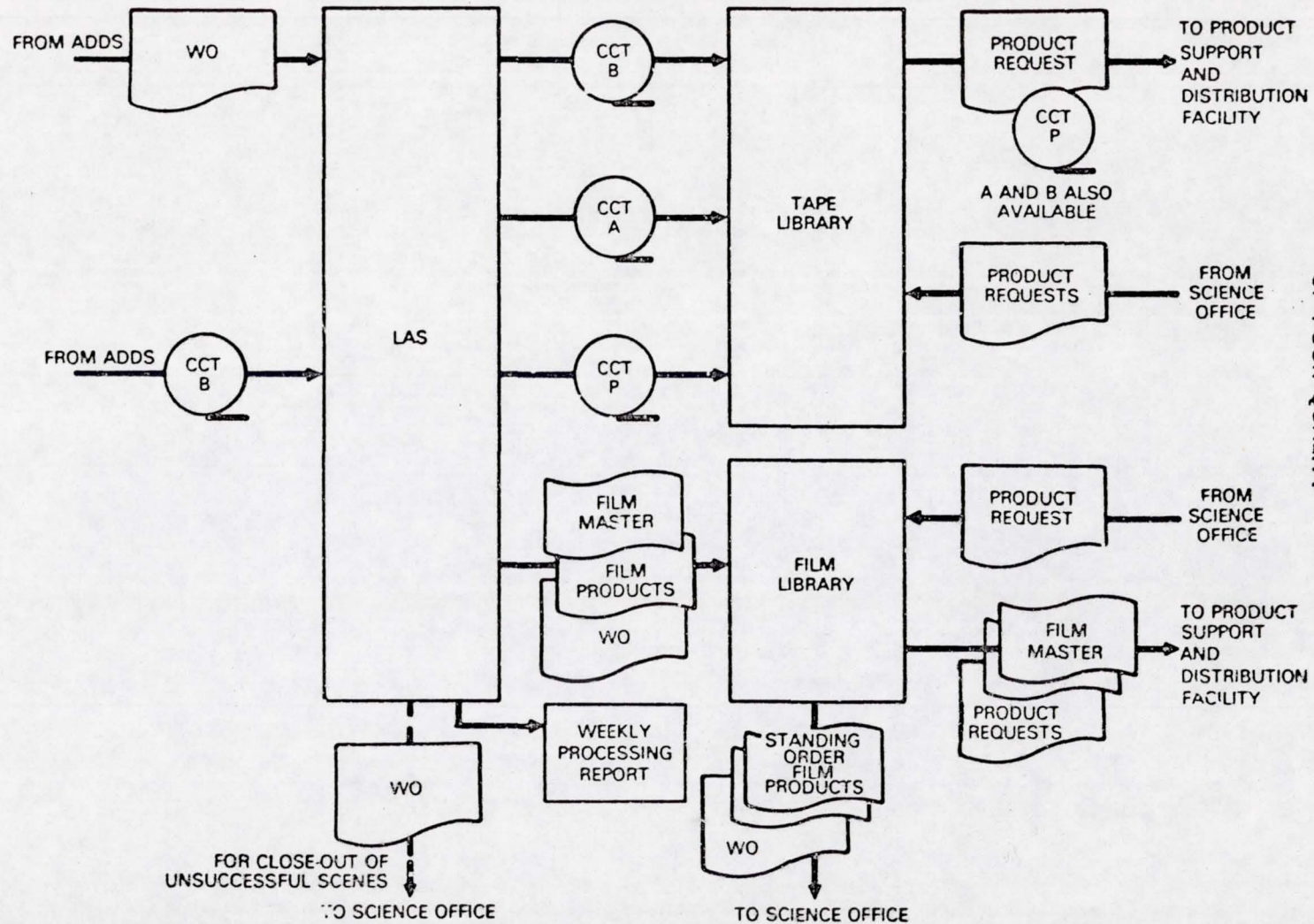
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REQUIRED PERFORMANCE LEVEL 1 TM SCENE/DAY

MSS CORRELATIVE DATA AVAILABLE: IGF AND IPF

MAJOR FUNCTIONAL DATA FLOW
- TM PROCESSING -

LAS interfaces



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Landsat-D Data Acquisition and Availability

Stan Freden

Landsat-D Data Acquisition and Availability

- **MSS Acquisition and Availability**
- **TM Acquisition and Availability**
- **AN Mission Options**

MSS Data Acquisition and Availability

FIRST YEAR

- Acquisition Capability:** 200 Scenes/Day
All US Except Hawaii
- Acquisition Priorities:** First 6 Months—All Possible Data
of US Consistent with Landsat-3/
Landsat-D Station Conflicts
After 6 Months—NOAA
Responsibility, NASA Engineering/
Special Requirements will be Met
- Acquisition Requests:** All Go Through the EROS Data
Center (EDC)
AN Requirements Come to GSFC
Science Office → EDC
- Processing Capability/Priority:** 200 Scenes/Day Soon After Launch
All Data Processed and Distributed
by EDC
Some Engineering Data Available
Through GSFC

Priorities for Loading Landsat-D MSS GCP Library (U.S.)

<u>AREA</u>	<u>PATH/ROW</u>
Washington, DC	15/33
Lubbock, TX area	29, 30, 31/36, 37
Grand Canyon, AZ	37, 38/35
Phoenix, AZ	37/37
Lake Powell, UT	37/34
San Francisco, CA	44/34
Los Angeles, CA	41/36
Chicago, IL	22, 23/31
White Sands, NM	33/37
Pennsylvania (especially Lancaster area)	(15/32)
Florida	
Midwest Agricultural Area (Iowa, Kansas, North Dakota, etc.)	
Eastern United States	
Western United States	
Rest of United States (48 states)	
Alaska	
Hawaii	

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MSS Tape Products

FACILITY	PROCESSING LEVELS				
	NONE	RADIOMETRIC		GEOMETRIC	
	RAW	INTERNAL CALIBRATION	SCENE HISTOGRAM	SYSTEMATIC	GEODETIC
MIPS CCT-AM CCT-PM	X	X	X	X	NOTE
EDC CCT-PM			X	X	NOTE

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Note: Available as Soon as GCP Library can be Loaded

TM Data Acquisition and Availability

FIRST YEAR

Acquisition Capability: 30 Scenes/Day Average

Eastern and Central 48 Prior to TDRSS

Total 48 + Some Alaska/Hawaii and Some
Foreign After TDRSS

Acquisition Priorities: Disasters

A/N Requirements

US Agricultural Requirements

Other Specials/PAO, etc.

Acquisition Requests: Data Request Forms

Inputs to Science Office

— AN Requirements go to Technical Representatives

— Others to Dr. Stanley C. Freden

NASA/Goddard Space Flight Center

Code 902

Greenbelt, MD 20771

344-5818

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LANDSAT-D

DATA REQUIREMENTS

THEMATIC MAPPER (TM)
AND
MULTISPECTRAL SCANNER (MSS)

PRINCIPAL INVESTIGATOR NAME _____

ADDRESS _____

TELEPHONE NUMBER (US AND CANADA) _____

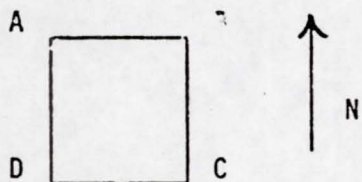
INVESTIGATION NUMBER _____

TEST SITE NUMBER _____

LOCATION _____

COORDINATES:

1-123



	LATITUDE D.M.S.	LONGITUDE D.M.S.
A.		
B.		
C.		
D.		

LANDSAT-D WRS
(Do not complete)

PATH	ROW

DATA ACQUISITION SCHEDULE:

DAY

NIGHT

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NIGHT JUSTIFICATION: _____

FIRST YEAR: (82) JUL AUG SEP OCT NOV DEC

(83) JAN FEB MAR APR MAY JUN

SECOND AND THIRD YEARS: (83/84) JUL AUG SEP OCT NOV DEC

(84/85) JAN FEB MAR APR MAY JUN

CLOUD COVER RESTRICTION: _____ %

ARE IN SITU DATA BEING COLLECTED CONCURRENTLY? YES NO

ANTICIPATED DATE(S) _____

DATA PRODUCTS:

THEMATIC MAPPER:

BAND	TAPES			FILM	
	CCT-A	CCT-P	+P	+T	-T
1					
2					
3					
4					
5					
6					
7					

1-124

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MULTISPECTRAL SCANNER:*

BAND	TAPES			FILM	
	CCT-A	CCT-P	+P	+T	-T
1					
2					
3					
4					

1-125

* The Principal Investigator must order Multispectral Scanner data from the ERUS Data Center. These data will be paid for by the Principal Investigator.

TM Data Acquisition and Availability (Cont.)

FIRST YEAR

Processing Capability = 1 Scene/Day = 7 Scenes/Week

Processing Priorities: Disasters
A/N Requirements—Commonality, etc.
US Agricultural Programs
Other Specials

**Processing Selection
and Data Distribution:** Screen and Select 21 "Cloud Free",
"High Quality" Scenes Each Week in
Priority Order
Produce = 10 CCT-B's in ADDS for LAS
Produce 7 "Processed" Scenes/Week in
LAS

Provide Tape and Film Copying
Provide Tape and Film Distribution

Expected Data \approx 1 Scene/Cycle/AN Average
Product Availability: Tapes and Film as Required
Availability of First Processed Scene Upon
Request for Analysis Systems Checkout

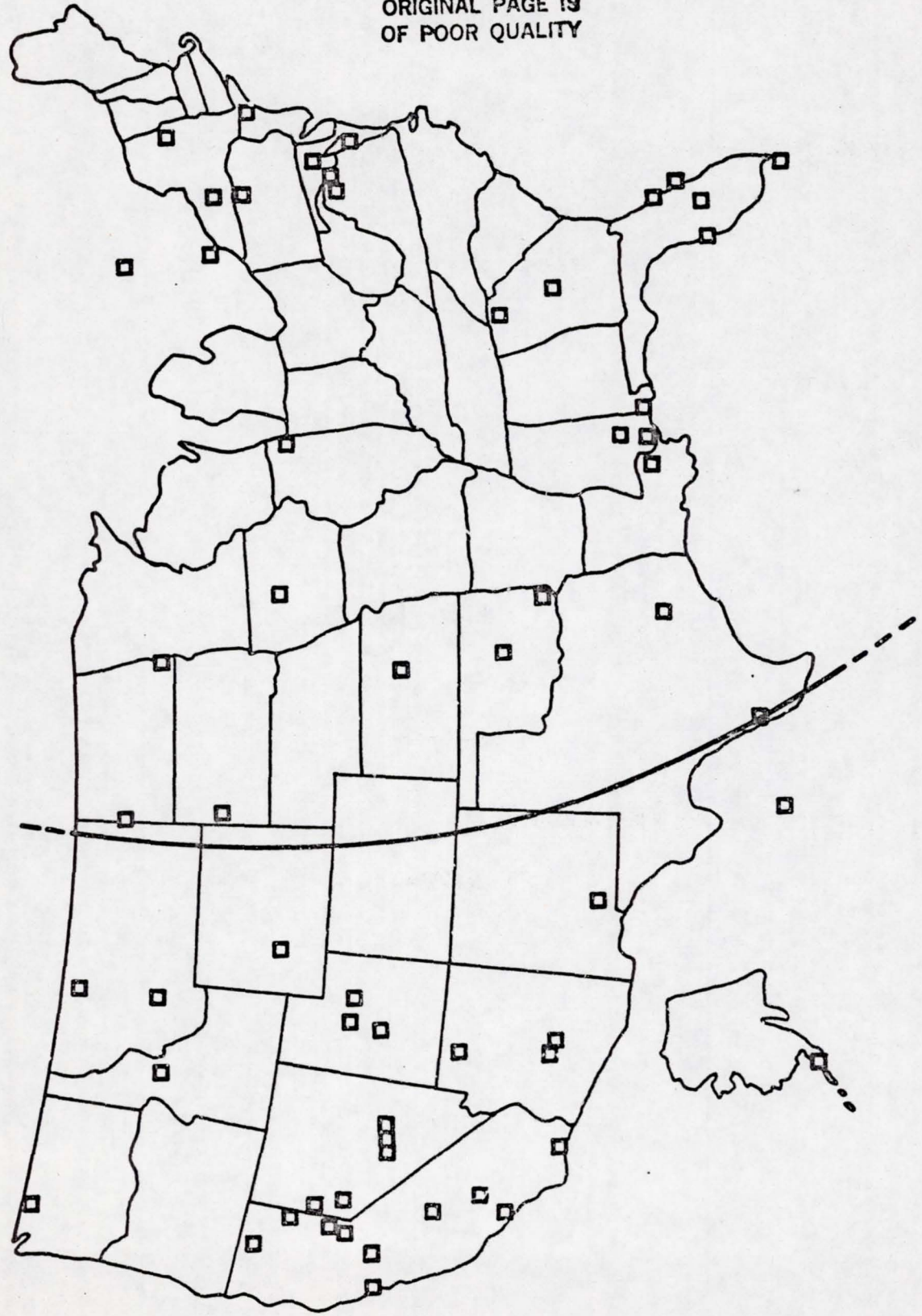
TM Tape Products

FACILITY	PROCESSING LEVELS				
	NONE	RADIOMETRIC		GEOMETRIC (NN OR CC RE-SAMPLING)	
	RAW	INTERNAL CALIBRATION	SCENE HISTOGRAM	SYSTEMATIC	GEODETIC
SCROUNGE (BEFORE JULY 83) CCT-BT CCT-AT CCT-PT	X	X X	X X	X	NOTE
TIPS CCT-AT CCT-PT	X	X X	X X	X	NOTE

Note: Available as Soon as GCP Library can be Loaded

Landsat-D Provisional Investigator Test Sites

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Examples of AN Landsat-D Mission Options

- MSS On Alone
- TM On Alone
- MSS and TM On Together
- Daytime and Nighttime
- Choice of MSS Configuration
- Choice of TM Configuration

Introduction of Technical Experts and Science Representatives

John Barker

**Introduction to MSS
Pre-NOAA Characterization**

Bill Alford

Landsat-D Science Office Pre NOAA MSS Characterization

OBJECTIVES

- Characterize Accuracy and Precision of Imagery
- Characterize Accuracy and Precision of Derived Information
- Recommend Landsat-D System Improvements
- Communicate Capabilities to Research Community

Landsat-D Science Office Pre NOAA MSS Characterization

AREAS OF INVESTIGATION

- Sensor Performance
- Image Data Quality
- Applications Information

SUPPORT

- Landsat-D Project
 - GE
 - Hughes Aircraft
 - CSC
 - ORI
- GSFC MSS Characterization Support
- Application Notice Investigations

Sensor and Spacecraft Performance Characterization

Radiometry of MSS

		Anuta	Bender	Slater	Malila										
Spectral Resolution	Spectral Matching	Filter													
		Detectors													
		System		●	●	●									
Radiometric Resolution	Absolute Integrating Sphere Calibration														
	External Calibration		●	●	●	●									
	Internal Calibration	Precision				●									
		Signal-to-Noise	●	●		●									
	Flooding Lamp Calibration														

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Sensor and Spacecraft Performance Characterization

Geometry of MSS

		Bernstein	Anuta	Bender	Keiffer																	
Geometry of Image (Pixel Location)	Sensor Effects	Scan Profile, Reference Detector	●	●	●	●																
		Detector Location Relative to Reference Detector		●	●	●																
		Between Scan Alignment		●		●																
	Ephemeris	Orbital Support Competing Div.																				
		Global Positioning System (GPS)																				
	Attitude	Angular Displacement Sensor (ADS)																				
		Inertial Reference System (DRIRU)																				
		Attitude Control System (ACS)																				
		Alignment to Sensor																				

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Sensor and Spacecraft Performance Characterization

Geometry of MSS (con't)

		Colewell	Anuta	Zobrist														
Geometry of Pixel	Rise Time & Delay Time	●																
	Bright Target Recovery Time	●																
	MTF (IFOV) or Frequency Response Time	●																
	Bowtie Scan Angle Effect	●	●															
	Altitude Effects	●	●	●														

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Image Data Quality Performance Characterization

Radiometry of MSS

		Colewell	Bernslein	Anuta	Bender	Slater	Malila	Zobrist	Hovis				
Spectral Information	Detector Replacement Algorithms				●								
	Band Compression Algorithms												
Radiometric Information	Internal Calibration Algorithms	Channel-to-Channel			●				●				
		Band-to-Band			●				●				
	Scene Histogram Calibration Algorithms (Radiometric Destriping)			●	●			●					
	Absolute Scene Radiance Calibration Algorithms	Reflective Band	●			●	●						
		Thermal Band	●				●						
	Noise Correction Algorithms				●				●				

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Image Data Quality Performance Characterization

Geometry of MSS

		Bernstein	Anuta	Berder	Wrigley	Mallia	Zobrist	Keiffer						
Geometry of Pixel	Ground IFOV	●												
Geometry of Image (Pixel Location)	Systematic Correction	Scan Profile	●	●										
		Detector Location	●	●	●				●					
		Between Scan Alignment												
		Ephemeris												
		Attitude		●										
	Geodetic Correction with GCPs	Reference Library Build	●		●	●	●		●					
		Scene-to-Scene Registration	●		●	●	●	●	●					
	Resampling													

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Applications Information – MSS and TM

Areas of Interest

RENEWABLE RESOURCES	NON-RENEWABLE RESOURCES	PLANNING/ENVIRONMENTAL MANAGEMENT
<p>Agriculture Inventory Yield Condition Irrigation Episodal Event</p> <p>Soils Classification Erosion Moisture</p> <p>Forests Inventory Stand Evaluation Condition Episodal Event</p> <p>Range Vegetation Inventory Condition Episodal Event</p>	<p>Geology Structure Landforms Lithology Thermal Anomalies Geobotanical Anomalies Topography (Stereo) Episodal Event</p> <p>Image-Science Pattern Recognition Information Extraction</p>	<p>Regional/Urban Land Use Cover Classification Cover Change Environmental Impact</p> <p>Coastal Zone Monitoring</p> <p>Hydrology Drainage Patterns Inland Water Inventory Snow Pack Parameters Ice – Inland & Near Shore Water Quality – Inland & Near Shore Wetland/Estuaries Inventory Episodal Event</p> <p>Wildlife Habitat Inventory Evaluation</p> <p>Oceans Currents (Near Shore) Tides Bathymetric Charts Ocean Pollution (Near Shore)</p>

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GSFC Radiometric Characterization

- **Destriping Analysis**
 - Detector Histogram Comparisons
 - Bright to Dark Area Comparisons
 - Visual, Clustering and Classification Qualitative Tests

- **Dynamic Range**

- **Signal-to-Noise**

GSFC Geometric Characterization

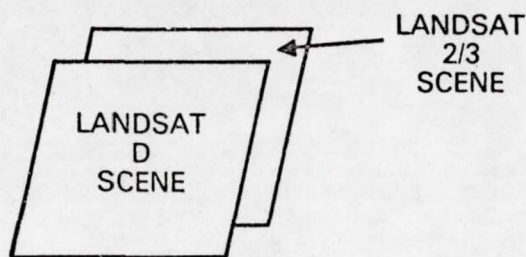
- **Geodetic Rectification Accuracy**
Compare with Intense Array of Verification Points

- **Temporal Registration Accuracy**
Cross Correlate Temporal Scene with Geodetic Verified Scene

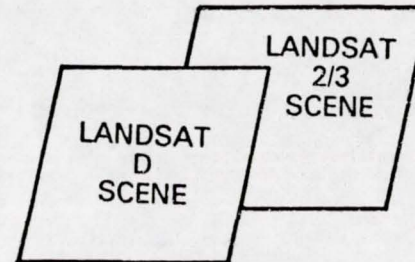
- **Systematic Correction Accuracy**
 - Cross Correlate With Geodetic Verified Scene
 - Band to Band Correlation to Measure Band Offsets
 - Analysis of MIPS Derived Parameters for Attitude/Ephemeris (GE)
 - Define Scan Non-Linearity (GE)

Landsat-D vs. Landsat 2/3 Geometric Accuracy

- Within Scene Comparisons as a Function of WRS Offset

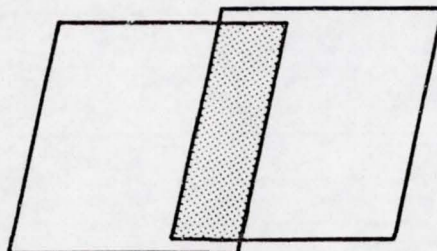


Minimum WRS
Offset

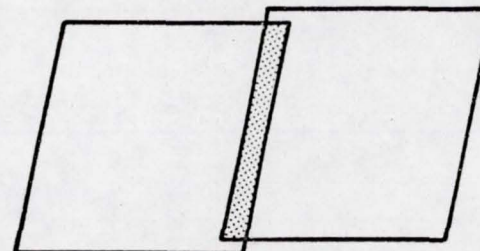


Maximum WRS
Offset

- Adjacent Scene Overlap Area Comparisons



Landsat-2/3
Overlap



Landsat-D
Overlap
(Smaller Than for 2/3)

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Landsat-D

Pre-NOAA MSS Characterization Reports & Schedules

REPORTS	1982												1983											
	J	F	M	A	M	J	J	A	S	O	N	D	J	F	M	A	M	J	J	A	S	O	N	D
Hughes MSS Final Report					○	△																		
GE Geometric Calibration Report									△															
GE Radiometric Calibration Report									△															
GE Geometric Evaluation Report																						△		
GE Radiometric Evaluation Report																						△		
CSC MSS Radiometric Calibration Report					○				△															
ORI MSS Geometric Correction Report					○				△															
GSFC MSS Characterization Report									○	△														
Applications Notice Investigations MSS Applications Information Report																						○	△	
Applications Notice Investigations MSS Characterization Report																						○	△	

Final = △
Draft = ○

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MSS Radiometric Sensor Performance

- **Spectral Information**
- **Absolute Calibration**
- **Ground Segment**

John Barker

LANDSAT-D MSS RADIOMETRY

OBJECTIVES

TODAY

DOCUMENT SPECTRAL CHARACTERISTICS

**DOCUMENT ABSOLUTE RADIOMETRIC LAMP
CALIBRATION**

**DOCUMENT POST-LAUNCH RADIOMETRIC
PREPROCESSING PROCEDURE**

FUTURE

RECALIBRATE AND VALIDATE POST-LAUNCH

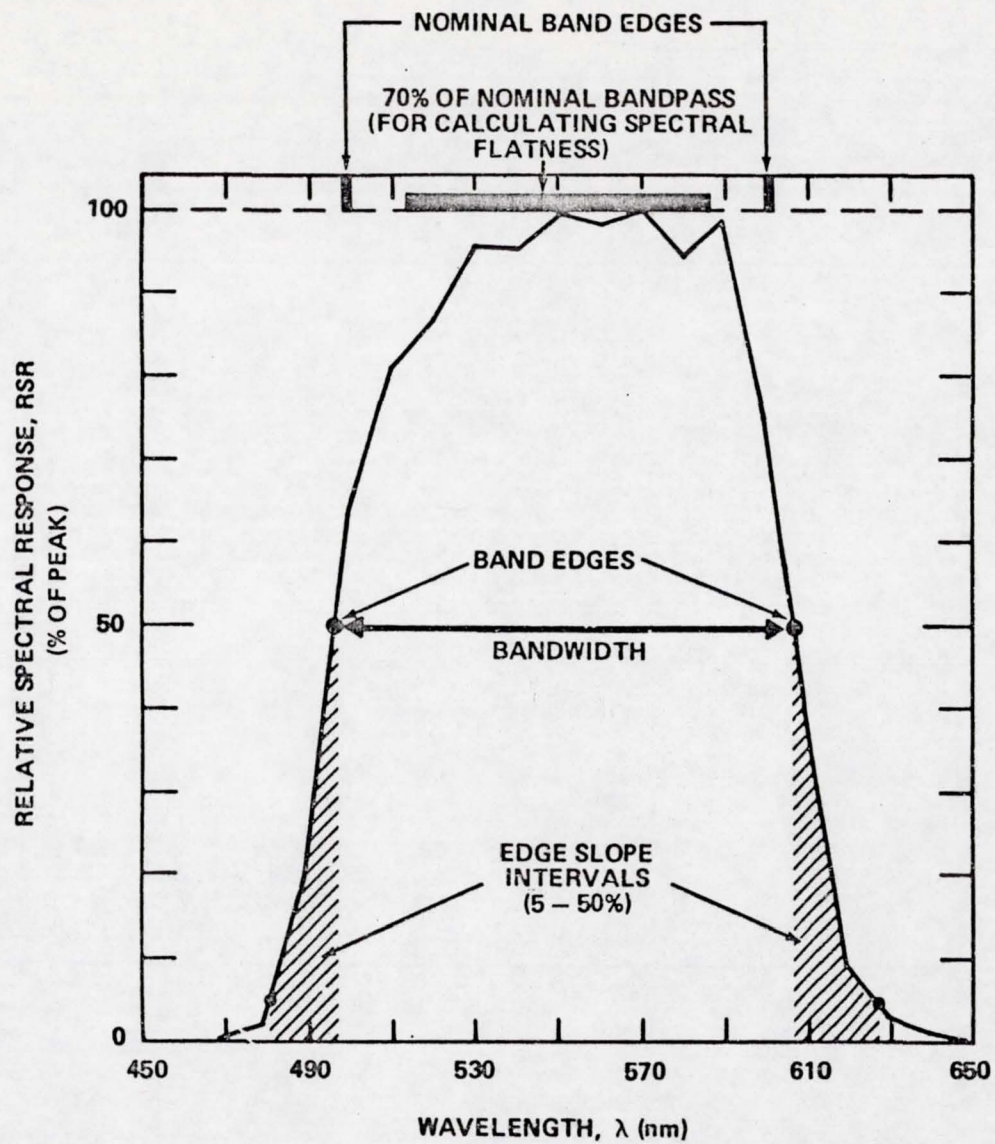
**ESTIMATE THE ACCURACY AND PRECISION OF
RADIOMETRIC CALIBRATIONS**

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FILTER SPECIFICATIONS FOR LANDSAT MULTISPECTRAL SCANNERS

BAND	BAND EDGE (nm)		BAND WIDTH (nm)	SLOPE INTERVAL (nm)		SPECTRAL FLATNESS (%)	
	HALF POWER POINTS LOWER	UPPER		FROM 5% TO 50% LOWER	UPPER	OVER CENTRAL 70% POSITIVE	NEGATIVE
1	500 ± 10	600 ± 10	—	<20	<40	<5.0	<5.0
2	600 ± 10	700 ± 10	—	<20	<45	<7.5	<7.5
3	700 ± 10	800 ± 10	—	<20	<50	<5.0	<5.0
4	800 ± 10	1100 ^a ± 10	—	<35	—	<5.0	<5.0

a — UPPER BAND EDGE NOT FILTER DETERMINED — FILTER SPECIFICATION
NECESSARY FOR FLATNESS DETERMINATION



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KEY TO FIGURES 2 AND 3

**MSS-D SPECTRAL CHARACTERIZATION BY
CHANNEL: BAND 2 (600-700 nm)**

	SCANNER CHANNEL	BAND EDGE (nm)		WIDTH ^a (nm)	SLOPE INTERVAL (nm)		SPECTRAL FLATNESS	
		LOWER	UPPER		LOWER	UPPER	POSITIVE	NEGATIVE
PROTO- FLIGHT	7	603	708*	105*	12	19	8.2 ^b	17.2 ^b
	8	602	696	94	12	16	6.4	11.6 ^b
	9	603	696	92	12	14	6.6	11.0 ^b
	10	603	696	94	12	18	7.8 ^b	11.1 ^b
	11	604	698	94	13	17	4.5	11.7 ^b
	12	602	695	93	12	15	8.2 ^b	14.5 ^b
FLIGHT	7	603	697	94	13	17	6.9	9.6 ^b
	8	603	696	93	13	16	7.3	10.4 ^b
	9	603	696	94	12	16	9.1 ^b	13.3 ^b
	10	602	696	93	12	14	9.1 ^b	16.0 ^b
	11	603	697	94	12	15	7.0	8.6 ^b
	12	603	697	94	12	15	6.4	8.5 ^b

a — NO FILTER SPECIFICATION

b — FAILS TO MEET FILTER SPECIFICATION

* — REJECTABLE AS OUTLIER: $\alpha = 0.01$

BAND 2 (600-700nm) SPECTRAL CHARACTERIZATION BY MEANS AND STANDARD DEVIATIONS: MSS 1, 2, 3, PF, F

	SCANNER	BAND EDGE (nm)		WIDTH ^a (nm)	SLOPE INTERVAL (nm)		SPECTRAL FLATNESS		
		LOWER	UPPER		LOWER	UPPER	POSITIVE	NEGATIVE	
MEANS	PF*	603	698	95	12	16	7.0	12.9 ^b	
	PF**	603	696	93	12	16	6.7	12.0 ^b	
	F	603	697	94	12	15	7.6 ^b	11.1 ^b	
	1	603	701	97	15	26	9.0 ^b	13.3 ^b	
	2*	607	710	103	14	30	7.9 ^b	18.0 ^b	
	2**	607	710	103	14	29	7.8 ^b	16.8 ^b	
	3	606	705	100	14	31	7.2	17.2 ^b	
	STANDARD DEVIATIONS	PF*	0.7	4.7	4.8	0.5	1.9	1.4	2.5
		PF**	0.8	0.8	0.6	0.5	1.4	1.5	1.4
		F	0.4	0.6	0.5	0.4	0.9	1.2	3.0
1		3.5	2.2	2.8	1.7	3.4	3.4	2.8	
2*		0.6	0.8	1.0	1.2	3.6	1.1	4.5	
2**		0.6	0.9	1.1	1.2	1.0	1.2	3.8	
3		0.9	1.2	0.8	0.8	2.0	2.0	4.8	

*WITH OUTLIER CHANNEL INCLUDED
**WITH OUTLIER CHANNEL EXCLUDED

a — NO FILTER SPECIFICATION
b — FAILS TO MEET FILTER SPECIFICATION

BOXES INDICATE CHARACTERISTICS WHERE DIFFERENCES BETWEEN PF OR F AND ALL PREVIOUS SCANNERS (1,2,3) WERE GREATER THAN DIFFERENCES BETWEEN TWO SETS OF PF MEASUREMENTS.

SIMULATED MSS BAND MEANS

TARGET	SENSOR SYSTEM	MEANS ^a (DIGITAL MSS COUNTS)			
		BAND 1 ^b	BAND 2 ^b	BAND 3 ^b	BAND 4 ^b
SOYBEANS	LSD-PF	19.36	14.89 (14.76) ^c	80.82*	45.80
	LSD-F	19.25	14.72	82.81*	45.39
	LS1	19.46 (19.55) ^c	15.43	76.95	47.14
	LS2	19.58	16.24 (16.13) ^c	78.58	47.24
	LS3	19.77	15.36	73.93	47.55
	LSD-PF	28.39	34.75 ^d	41.02	18.61
	LSD-F	28.39	34.75	41.05	18.48
SOIL	LS1	28.32 ^d	34.73	41.04	19.02
	LS2	28.34	34.66 ^d	41.05	19.07
	LS3	28.33	34.66	41.10	19.15

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a — AT SATELLITE SENSOR RESPONSE, NADIR.—LOOKING FOR 40° SOLAR ZENITH ANGLE AND 20 km VISIBILITY; UNITS ARE SIMULATED NON-TRUNCATED MSS DIGITAL COUNTS WITH MAXIMUM SPECIFIED RADIANCE SCALED TO 127.99 FOR BANDS 1, 2, 3 AND 63.99 FOR BAND 4.

b — LANDSAT-D BANDS 1, 2, 3 AND 4 CORRESPOND TO BANDS 4, 5, 6 AND 7, RESPECTIVELY ON PREVIOUS LANDSATS.

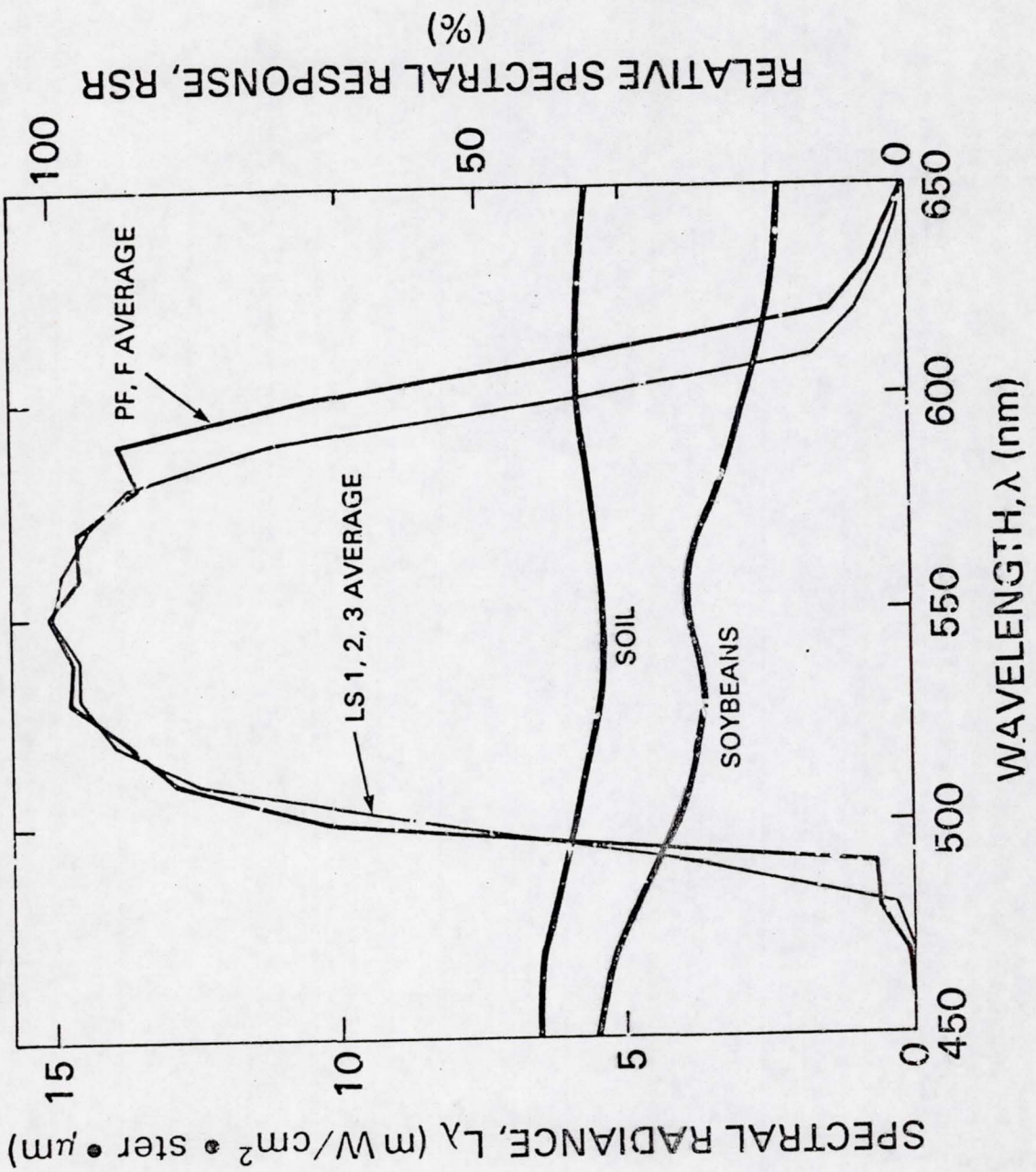
c — MEAN IN PARENTHESES IS WITH OUTLIER CHANNEL EXCLUDED

d — EXCLUSION OF OUTLIER DID NOT CHANGE BAND MEAN

* PF, F DIFFERENCE EXCEEDS: (1) DIFFERENCE BETWEEN SIMULATIONS RUN WITH EACH SET OF PF MEASUREMENTS SEPARATELY AND (2) 0.30 DIGITAL COUNTS

BOXES INDICATE BANDS WHERE OUTPUT DIFFERENCES BETWEEN PF OR F AND ALL PREVIOUS SCANNERS (1,2,3) EXCEED: (1) AND (2) AS ABOVE.

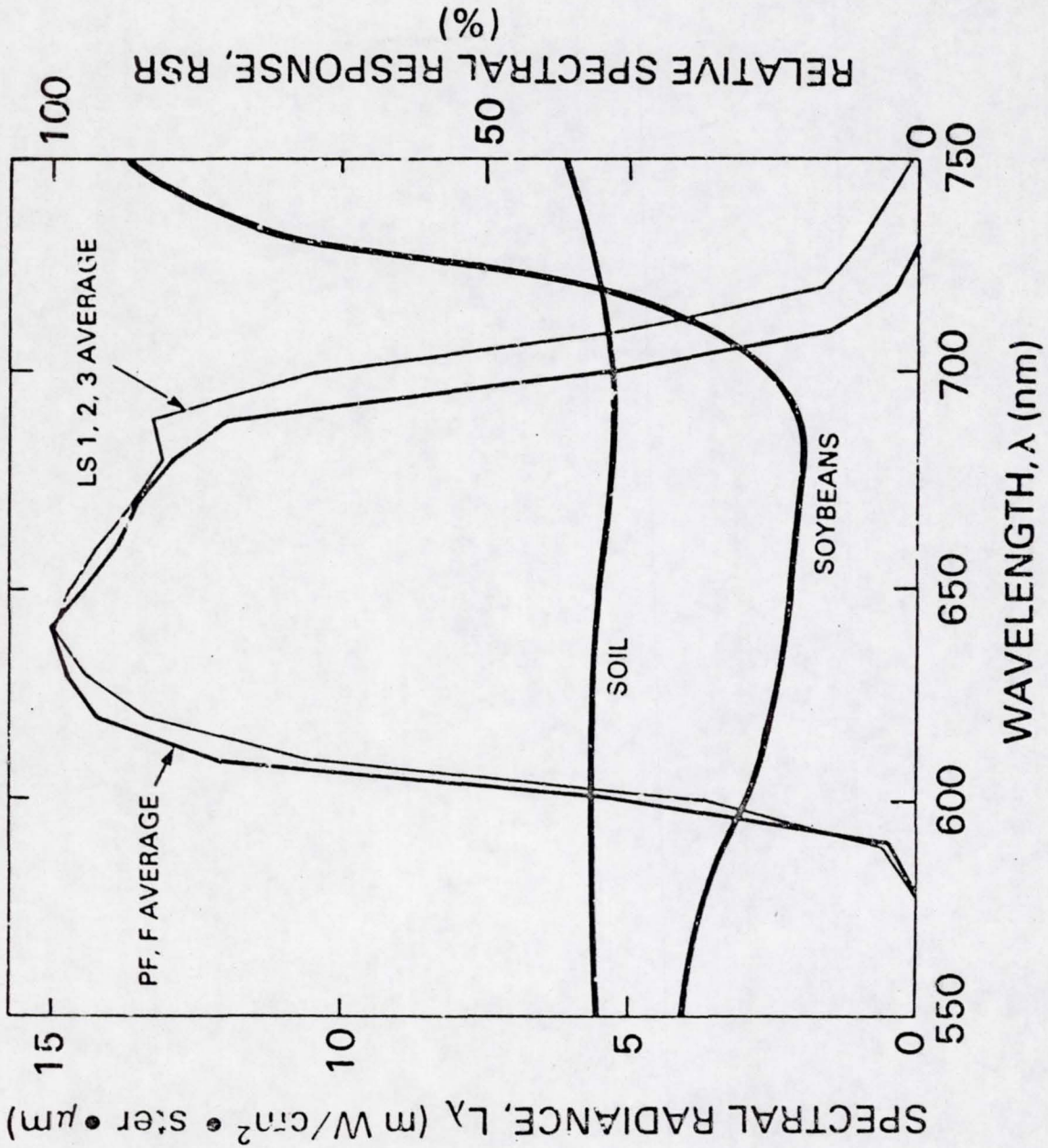
BAND 1 MSS-D DIFFERENCES



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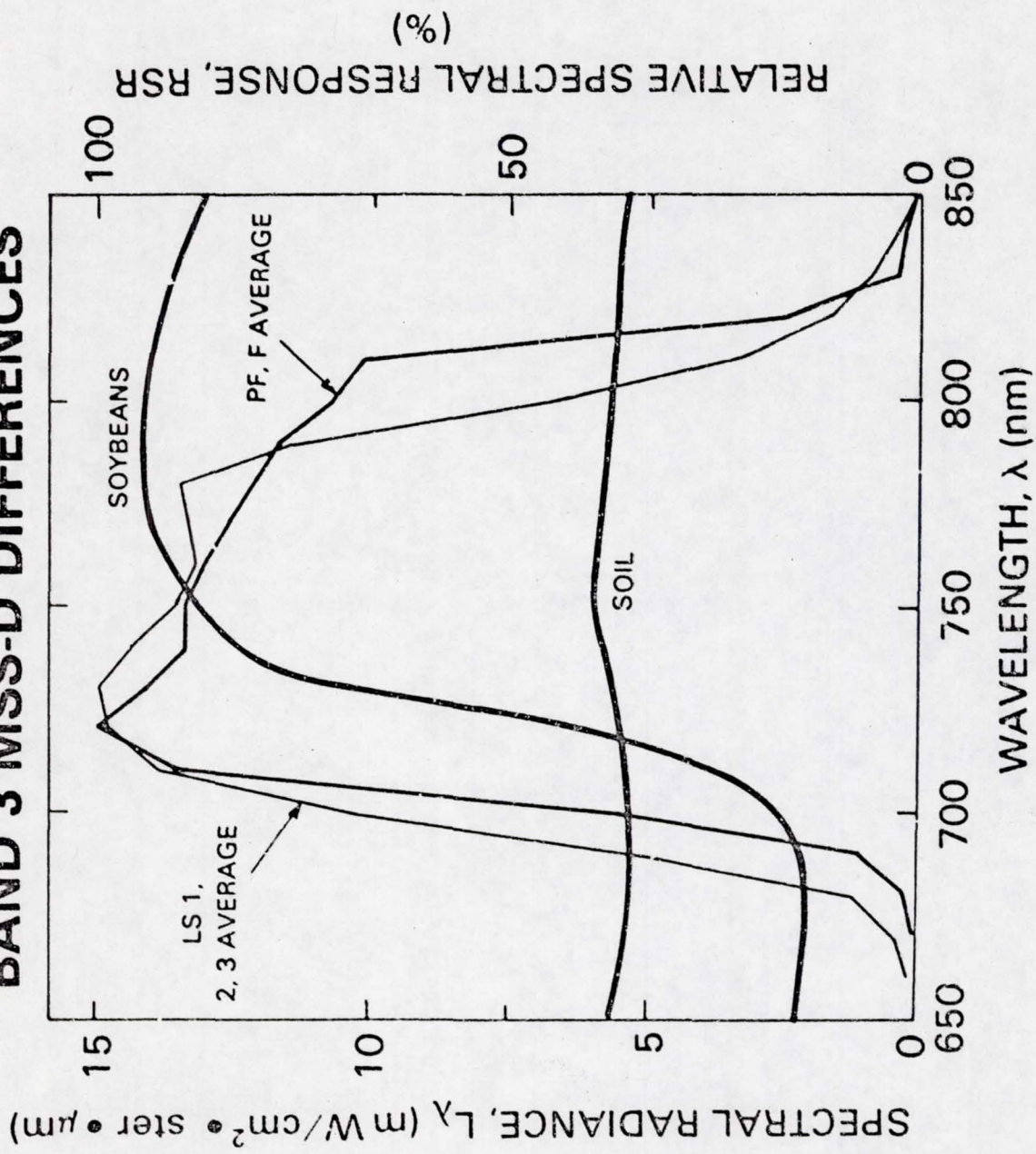
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BAND 2 MSS-D DIFFERENCES

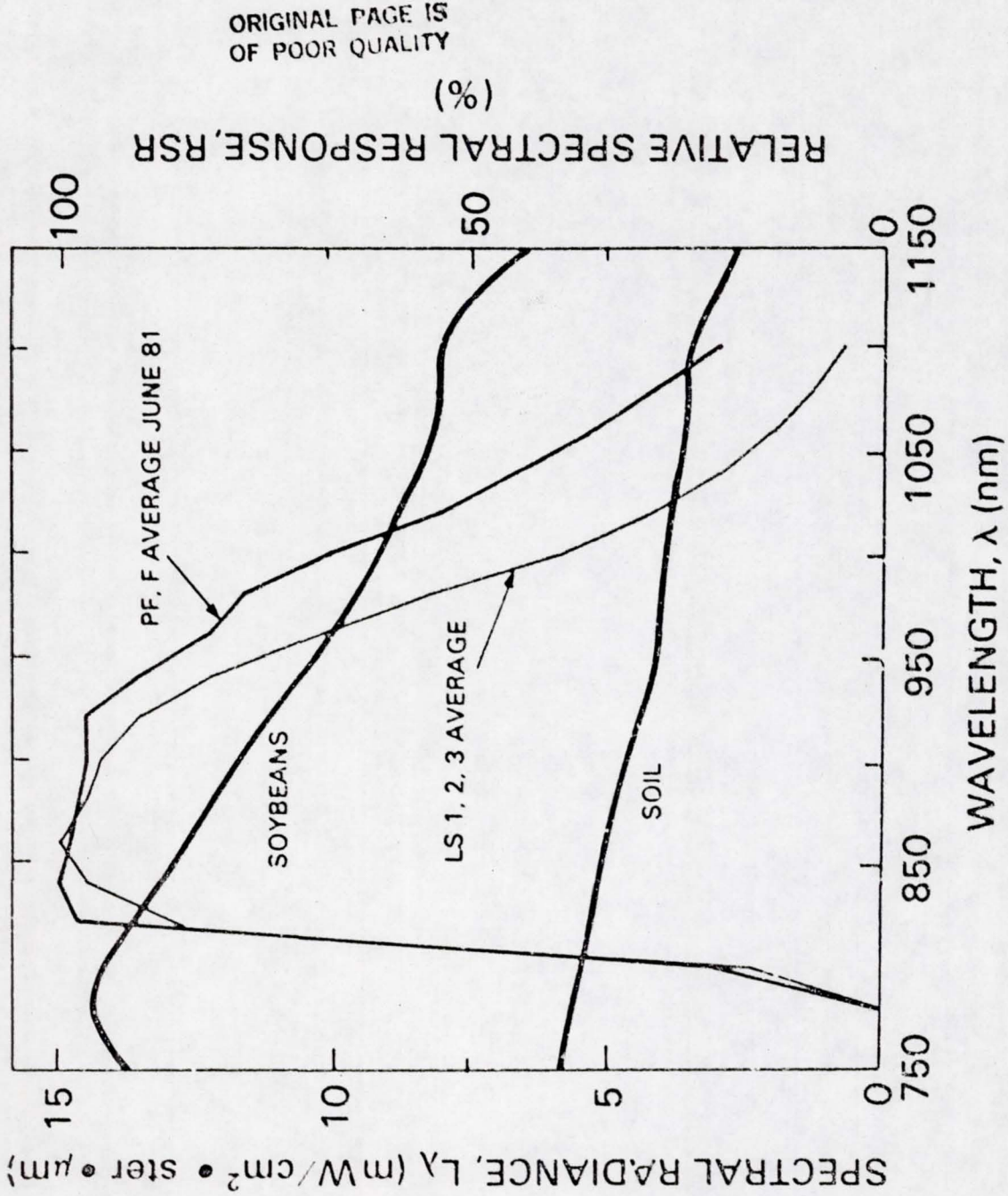


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BAND 3 MSS-D DIFFERENCES



BAND 4 MSS-D APPARENT DIFFERENCES



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(%)

RELATIVE SPECTRAL RESPONSE, RSR

100

50

0

1150

1050

950

850

750

WAVELENGTH, λ (nm)

15

10

5

0

SPECTRAL RADIANCE, L_λ ($\text{mW}/\text{cm}^2 \cdot \text{ster} \cdot \mu\text{m}$)

PF, F AVERAGE JUNE 81

SOYBEANS

LS 1, 2, 3 AVERAGE

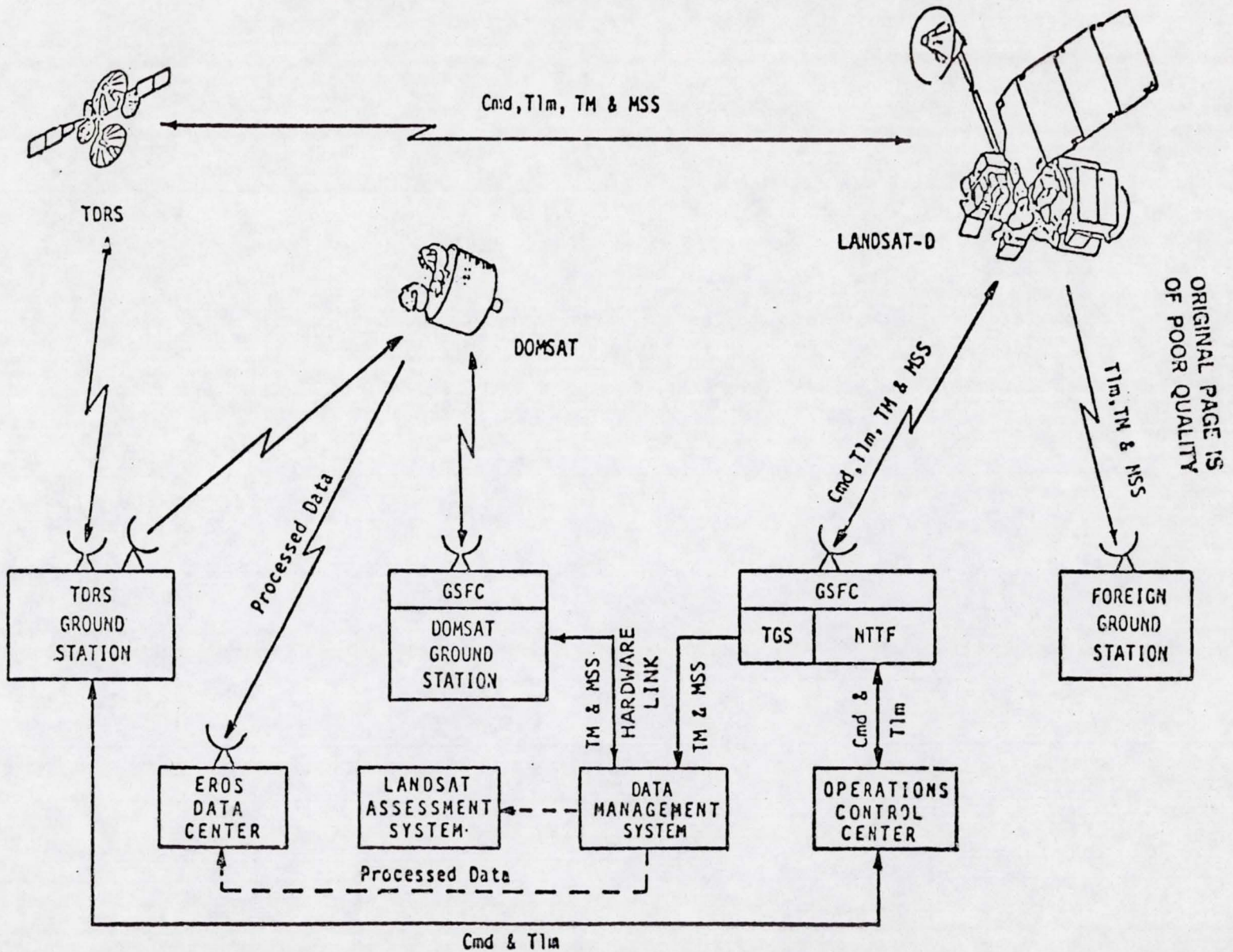
SOIL

**SIMULATED MAXIMUM WITHIN-BAND
SENSOR OUTPUT DIFFERENCES**

TARGET	SENSOR	DIGITAL COUNTS				PERCENT			
		BAND 1	BAND 2	BAND 3	BAND 4	BAND 1	BAND 2	BAND 3	BAND 4
SOYBEANS	LSD-PF	0.11	0.91*	2.23*	1.43*	0.6	6.2	2.8	3.1
	LSD-F	0.17	0.10*	0.78*	1.04*	0.9	0.7	0.9	2.3
	LS1	0.75	0.12	2.39	0.63	3.9	0.8	3.1	1.3
	LS2	0.16	0.77	3.63	0.39	0.8	4.8	4.6	0.8
	LS3	0.30	0.16	4.01	0.80	1.5	1.0	5.4	1.7
SOIL	LSD-PF	0.03	0.07	0.10	0.46	0.1	0.2	0.2	2.5
	LSD-F	0.01	0.05	0.02	0.32	0.1	0.2	0.1	1.8
	LS1	0.10	0.09	0.04	0.21	0.3	0.3	0.1	1.1
	LS2	0.05	0.03	0.06	0.12	0.2	0.1	0.2	0.6
	LS3	0.07	0.09	0.13	0.26	0.2	0.3	0.3	1.4

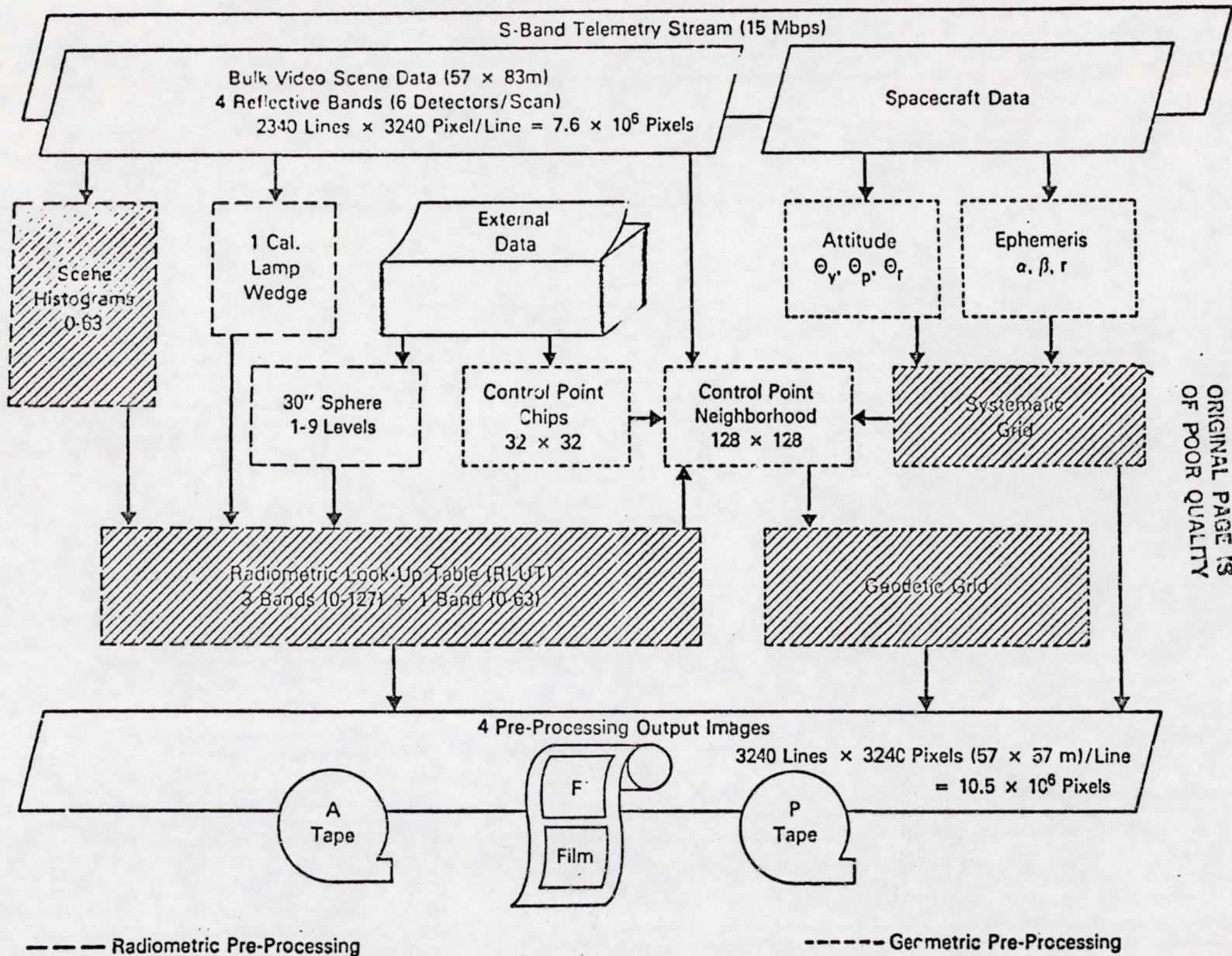
* PF, F DIFFERENCE EXCEEDS: (1) DIFFERENCE BETWEEN SIMULATIONS RUN WITH EACH SET OF PF MEASUREMENTS SEPARATELY AND (2) 0.30 DIGITAL COUNTS
BOXES INDICATE BANDS WHERE OUTPUT DIFFERENCES BETWEEN PF OR F AND ALL PREVIOUS SCANNERS EXCEED (1) AND (2) AS ABOVE

Landsat-D System Overview



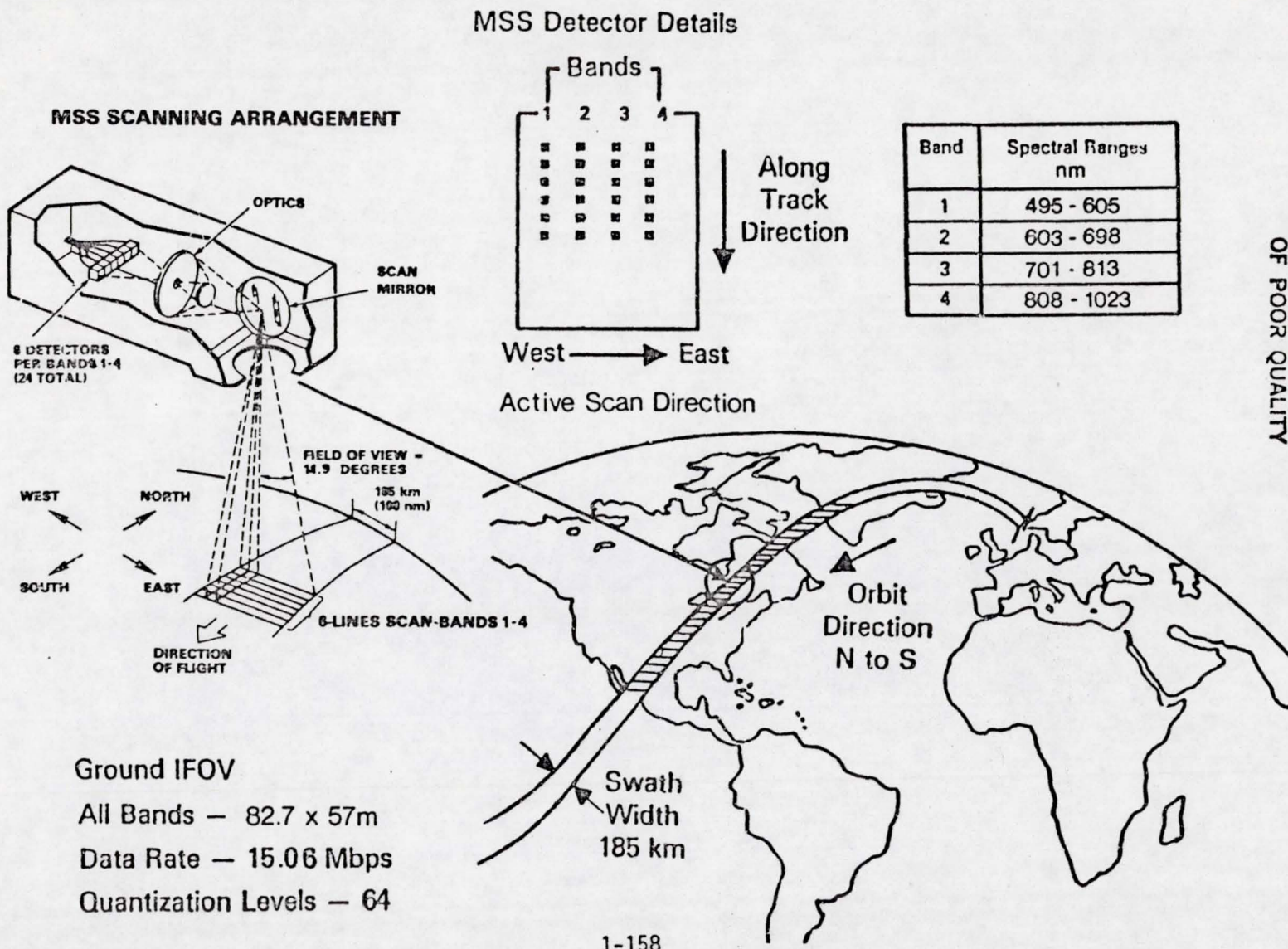
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MSS Landsat-D Preprocessing



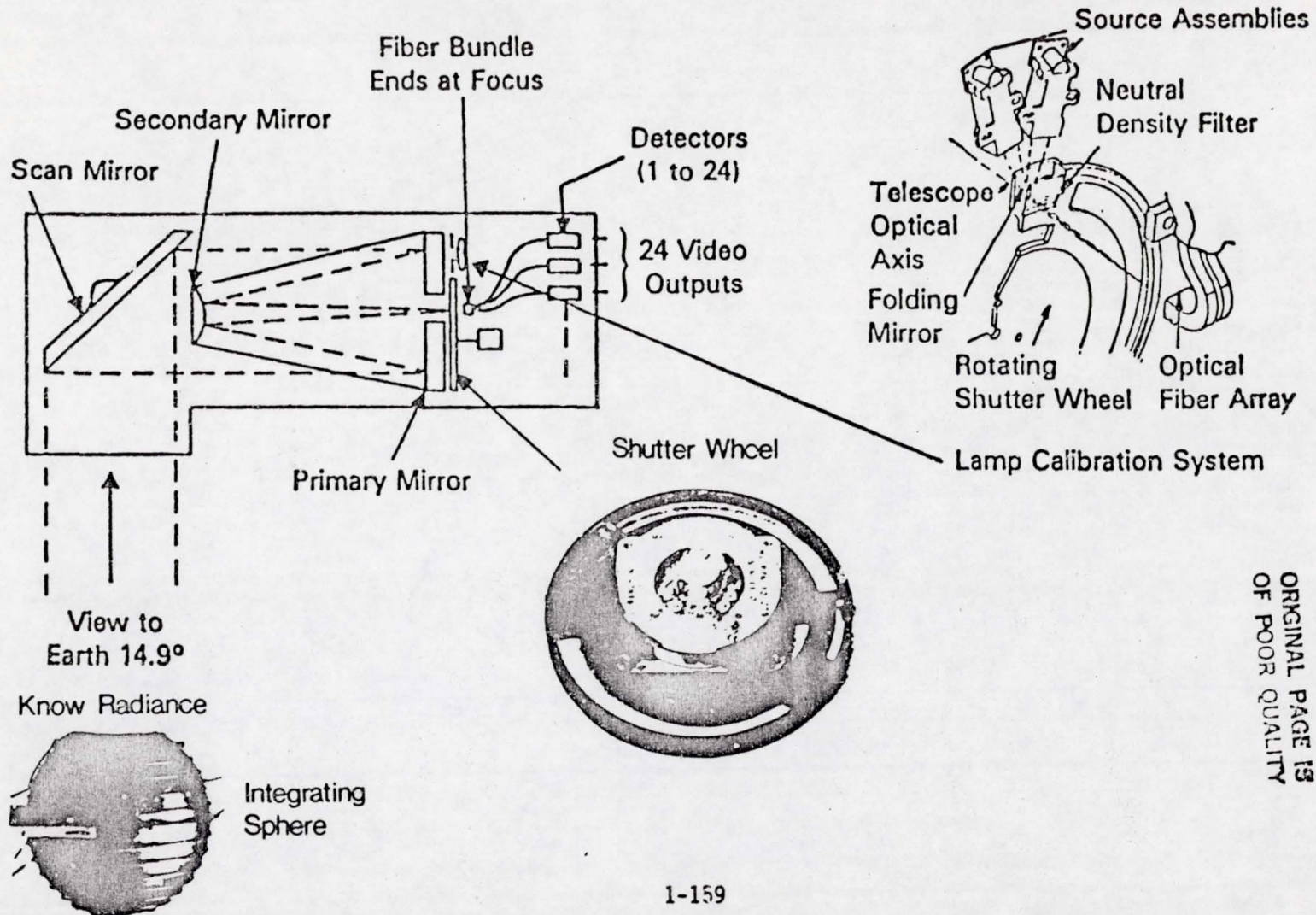
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Landsat-D Protoflight Multispectral Scanner (MSS)



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Landsat MSS Absolute Radiometric Calibration



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MSS CALIBRATION: DEFINITION OF VARIABLES

R_{JK} = INTEGRATING SPHERE RADIANCE FOR SPHERE LEVEL J
AND MSS CHANNEL K

$$R_{JK} = BW_K \frac{\int RW_J RSR_K d\lambda}{\int RSR_K d\lambda}$$

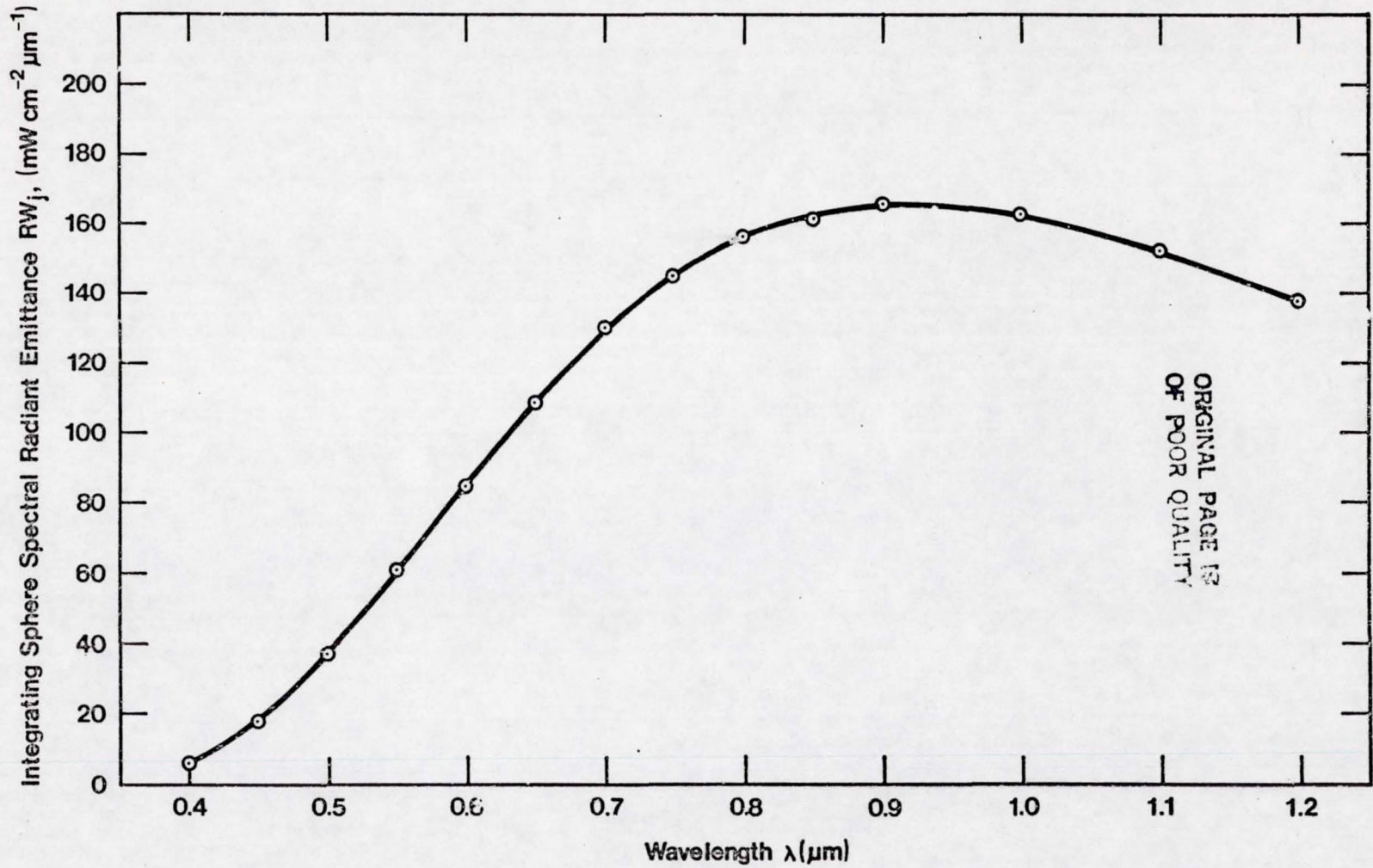
BW_K = MEASURED BANDWIDTH OF CHANNEL K

RSR_K = MEASURED RELATIVE SPECTRAL RESPONSE FOR CHANNEL K

RW_J = SPECTRAL RADIANCE FOR 76 CM INTEGRATING SPHERE FOR SPHERE LEVEL J

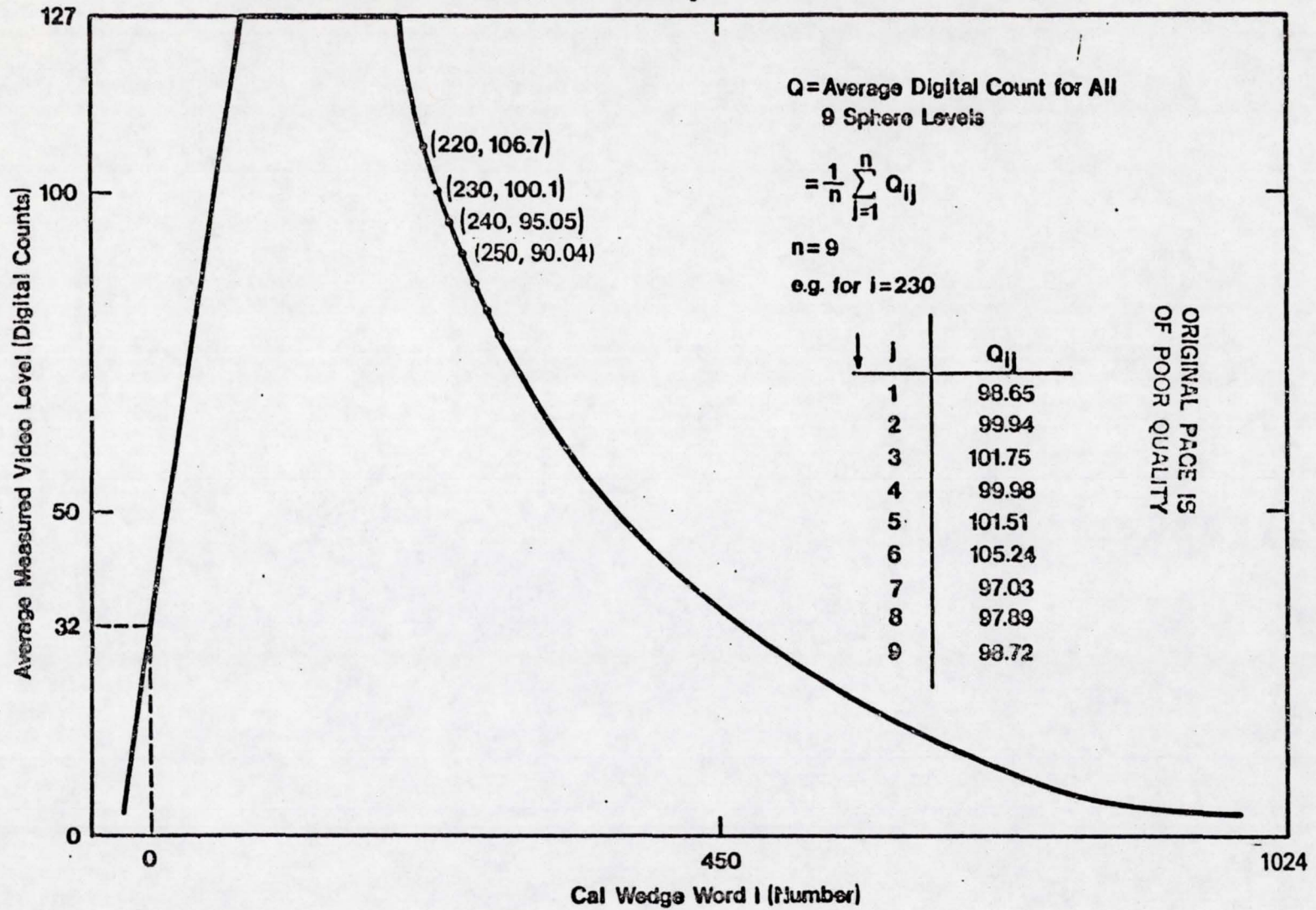
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Spectral Radiant Emittance Plot for 76 cm Integrating Sphere

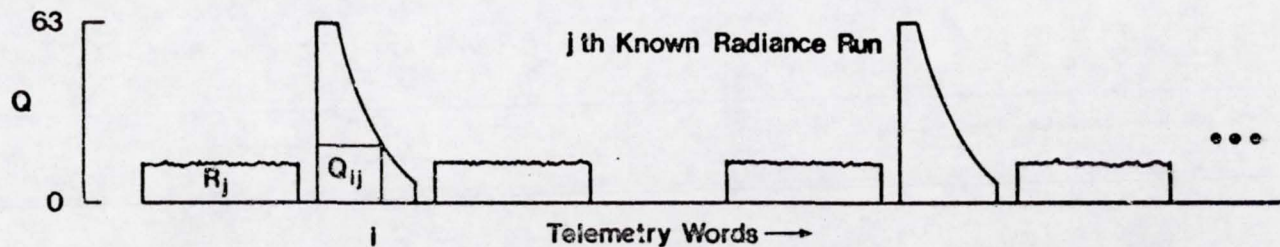
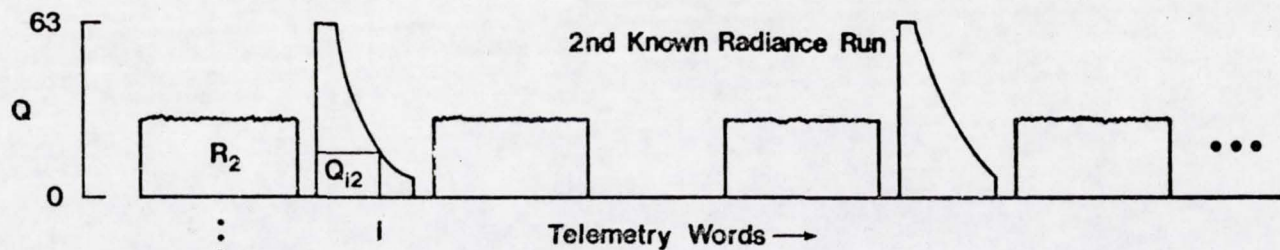
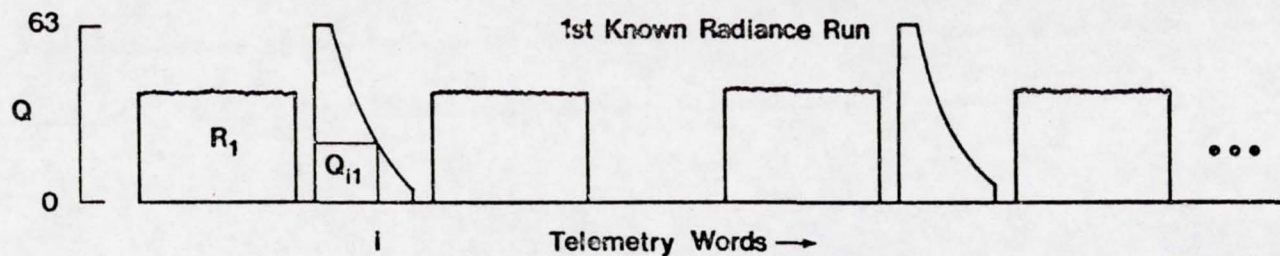


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Illustrative MSS/PF Lamp Calibration Wedge



Systematic MSS Video and Wedge Level Timing Sequence



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MSS CALIBRATION: DEFINITION OF VARIABLES (CONT'D.)

V_J = SCENE AVERAGES VIDEO LEVEL
FOR SPHERE LEVEL J

$$V_J = \frac{1}{390} \sum_{L=1}^{390} \frac{1}{N} \sum_{s=1}^N v_{JSL}$$

v_{JSL} = RAW VIDEO LEVEL
IN DIGITAL COUNTS
OF PIXEL s ON
LINE L FOR SPHERE
LEVEL J

s = INDEX FOR SUM OF
PIXELS IN A LINE

L = INDEX FOR SUM
OF LINES IN A
SCENE

Q_{IJ} = AVERAGE WEDGE LEVEL AT
WORD i AFTER SPHERE
LEVEL J

$$Q_{IJ} = \frac{1}{390} \sum_{L=1}^{390} q_{IJL}$$

q_{IJL} = RAW DIGITAL COUNTS FOR
WEDGE WORD i ON LINE
 L AFTER SPHERE LEVEL J

i = WORD NUMBER

J = SPHERE RADIANCE LEVEL

L = LINE NUMBER

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MSS/PF ILLUSTRATIVE ABSOLUTE CALIBRATION TRANSFER
 FROM SPHERE TO CALIBRATION LAMP WEDGE WORD (230)
 FOR BAND 1, CHANNEL 1
 (15TH SEPT., 1981, AT GE, VALLEY FORGE)

INTEGRATING SPHERE RADIANCE LEVEL	AVERAGED VIDEO LEVEL	AVERAGED WEDGE LEVEL (390 SCANS)	
R_J	V_J	Q_{iJ}	
(MW CM ⁻² SR ⁻¹)	(DIGITAL COUNTS)	(DIGITAL COUNTS)	
.04	2.1	97.03	ORIGINAL PAGE IS OF POOR QUALITY
.09	3.5	97.89	
.12	5.5	98.72	
.17	7.7	98.65	
.33	15.4	99.94	
.48	22.2	101.75	
.64	29.4	99.98	
.96	46.6	101.51	
1.95	96.6	105.24	

$$\bar{Q}_1 = \frac{1}{9} \sum_{j=1}^9 Q_{1j} = 100.08$$

ILLUSTRATES HYSTERESIS DEPENDENCE OF WEDGE VALUE ON PRECEDING RADIANCE LEVEL

CORRECTION PROCEDURE FOR HYSTERESIS EFFECT

- MODEL EXPECTED CAL WEDGE VALUE FOR WORD i AFTER SPHERE LEVEL J AS A FUNCTION OF AVERAGE VALUE FOR ALL SPHERE LEVELS

$$Q_{iJ} = A_J + B_J \bar{Q}_i$$

- ADJUST VIDEO VALUE FOR SPHERE LEVEL J ASSUMING THE HYSTERESIS TIME CONSTANT IS LONG RELATIVE TO THE 73 MSEC SCAN TIME

$$V_J = A_J + B_J VA_J$$

$$VA_J = \frac{V_J - A_J}{B_J}$$

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LINEAR HYSTERESIS MODEL FOR LAMP
 CALIBRATION WEDGE VALUES FOR MSS/PF
 BAND 1, CHANNEL 1

I	\bar{Q}_I (DIGITAL COUNTS)	Q_{IJ} (DIGITAL COUNTS)
220	106.7	112.9
230	100.1	105.2
240	95.1	99.7
250	90.0	94.6
260	85.7	90.7
270	81.3	85.9
280	77.5	81.6
290	73.0	77.1
300	68.9	73.2
310	65.5	68.7
810	4.1	4.3
820	3.9	4.1
850	3.2	3.3
860	3.0	3.2

LEAST SQUARES
 FIT TO

$$Q_{IJ} = A_J + B_J \bar{Q}_I$$

FOR BRIGHTEST SPHERE
 LEVEL J WITH

$$R_J = 1.95 \text{ MW CM}^{-2} \text{ SR}^{-1}$$

HYSTERESIS OFFSET

$$A_J = .02$$

HYSTERESIS GAIN

$$B_J = 1.054$$

$$R^2 = .99996$$

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ADJUSTMENT OF SPHERE VIDEO LEVELS FOR
HYSTERESIS EFFECT FOR MSS/PF BAND 1, CHANNEL 1

AVERAGED VIDEO VALUE FOR INTEGRATING SPHERE LEVEL J V_J (DIGITAL COUNTS)	LINEAR REGRESSION COEFFS $Q_{IJ} = A_J + B_J \bar{Q}_I$		ADJUSTED VIDEO LEVEL $VA_J = \frac{V_J - A_J}{B_J}$ (DIGITAL COUNTS)	SPHERE RADIANCE R_J ($MWCM^{-2}SR^{-1}$)
	A_J	B_J		
2.1	.01	.971	2.15	.04
3.5	-.08	.980	3.65	.09
5.5	-.01	.984	5.60	.12
7.7	.00	.986	7.81	.17
15.4	.00	1.000	15.39	.33
22.2	.01	1.014	21.89	.48
29.4	.03	.997	29.47	.64
46.6	.01	1.014	45.96	.96
96.6	.02	1.054	91.61	1.95

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CALIBRATION PROCEDURE FOR LAMP RADIANCE
TRANSFER OF SPHERE RADIANCE TO DETECTOR VIDEO
VALUE

LEAST SQUARES FIT: $VA_J = p + qR_J$

MSS/PF BAND 1, CHANNEL 1, SPHERE TO DETECTOR

OFFSET $p = -.21$ (DIGITAL COUNTS)

GAIN $q = 47.17$ (DIGITAL COUNTS/MWCM⁻²SR⁻¹)

ASSUME LAMP RADIANCE CAN BE CALCULATED FROM p, q

$$\bar{Q}_I = p + qR_I$$

$$R_I = \frac{\bar{Q}_I - p}{q}$$

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CALCULATION OF LAMP RADIANCE FOR SIX OPERATIONAL
WEDGE WORDS FOR MSS/PF BAND 1, CHANNEL 1

WEDGE WORD NUMBER	OBSERVED AVERAGE WEDGE LEVEL	CALCULATED RADIANCE FOR WEDGE WORD
I	\bar{Q}_I	$R_I = \frac{\bar{Q}_I - P}{q}$
	(DIGITAL COUNTS)	(MWCM ⁻² SR ⁻¹)
230	100.08	2.126
240	95.05	2.020
250	90.04	1.913
260	85.67	1.821
810	4.07	.091
820	3.85	.086

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CALCULATION OF LAMP RADIANCE FOR SIX OPERATIONAL
WEDGE WORDS FOR MSS/PF BAND 1, CHANNEL 1

WEDGE WORD NUMBER	OBSERVED AVERAGE WEDGE LEVEL	CALCULATED RADIANCE FOR WEDGE WORD
I	\bar{Q}_I	$R_I = \frac{\bar{Q}_I - p}{q}$
	(DIGITAL COUNTS)	(MWCM ⁻² SR ⁻¹)
230	100.08	2.126
240	95.05	2.020
250	90.04	1.913
260	85.67	1.821
810	4.07	.091
820	3.85	.086

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INTERNAL LAMP RADIANCE VALUES FOR SIX
 CALIBRATION WORD LOCATIONS FOR MSS/PF
 BAND 1

CHANNEL	230	240	250	260	810	820
1	2.126	2.020	1.913	1.821	.091	.086
2	2.056	1.952	1.857	1.763	.085	.081
3	2.099	1.992	1.899	1.808	.089	.084
4	2.047	1.944	1.849	1.760	.086	.080
5	2.062	1.966	1.854	1.766	.089	.083
6	2.058	1.950	1.856	1.770	.087	.082

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CALCULATION OF GAIN AND OFFSET USING LAMP RADIANCE VALUES
AT SIX OPERATIONAL WEDGE WORDS

$$\text{LINEAR REGRESSION: } \bar{Q}_I = \alpha + \beta R_I$$

$$\text{CHANNEL OFFSET} = \alpha = \sum_{I=1}^6 C_I \bar{Q}_I$$

$$\text{CHANNEL GAIN} = \beta = \sum_{I=1}^6 D_I \bar{Q}_I$$

CALIBRATION REGRESSION COEFFICIENTS:

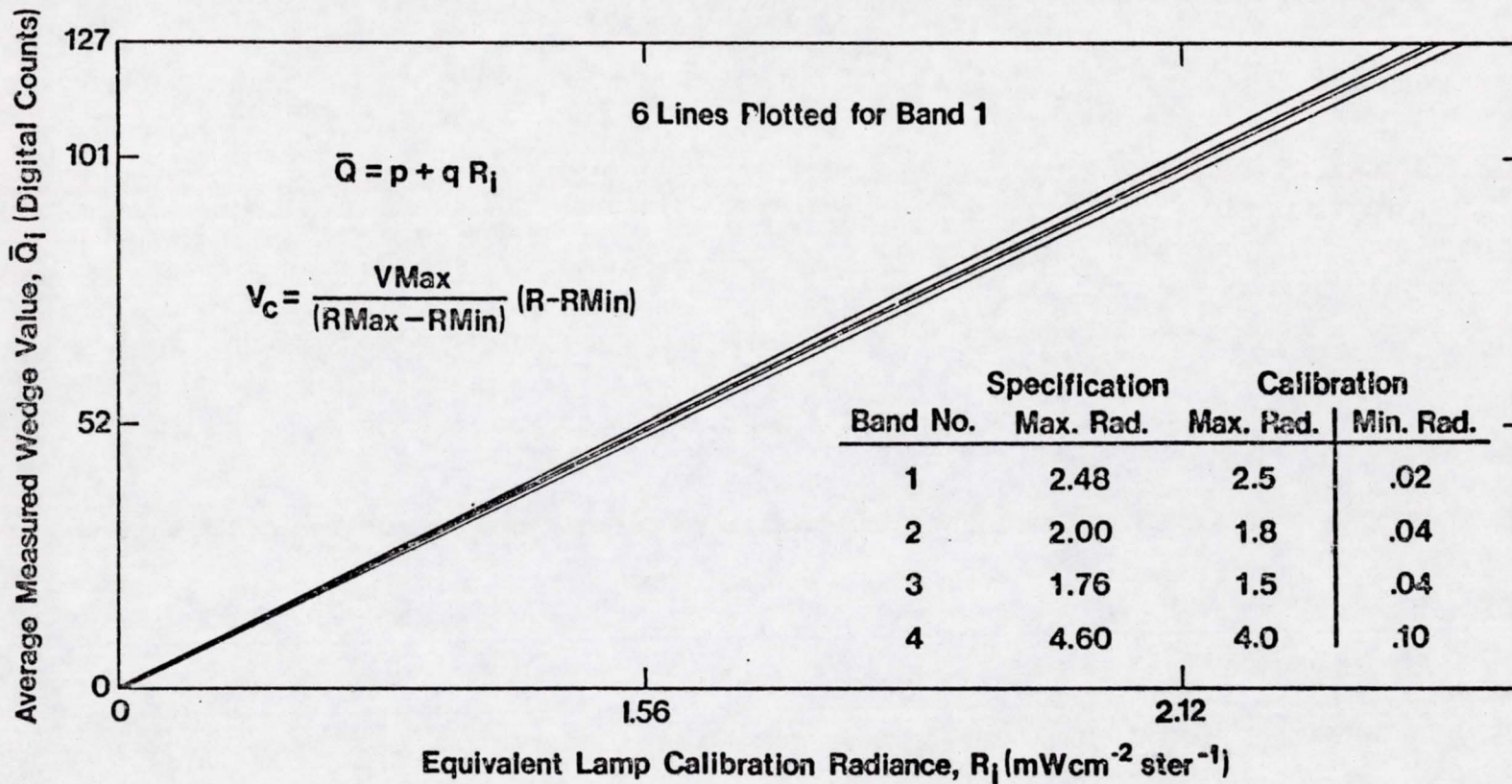
$$C_I = \left[\sum R_I^2 - R_I \sum R_I \right] / K$$

$$D_I = \left[6R_I - \sum R_I \right] / K$$

$$K = 6 \sum R_I^2 - (\sum R_I)^2$$

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BAND NORMALIZATION OF MSS CHANNEL GAINS AND OFFSETS



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ILLUSTRATIVE MSS/PF LINEAR REGRESSION COEFFICIENTS
FOR BAND 1

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WORD 230		WORD 240		WORD 250		WORD 260		WORD 810		WORD 820	
D ₁	C ₁	D ₂	C ₂	D ₃	C ₃	D ₄	C ₄	D ₅	C ₅	D ₆	C ₆
.1641	-.054	.1418	-.024	.1195	.006	.1002	.032	-.2624	.519	-.2633	.520
.1689	-.053	.1457	-.023	.1244	.005	.1035	.032	-.2707	.518	-.2718	.520
.1648	-.052	.1420	-.022	.1220	.005	.1026	.030	-.2651	.519	-.2662	.520
.1691	-.052	.1461	-.022	.1247	.005	.1048	.031	-.2717	.518	-.2730	.520
.1688	-.053	.1474	-.025	.1225	.007	.1029	.033	-.2700	.519	-.2716	.521
.1690	-.053	.1450	-.022	.1240	.005	.1047	.031	-.1707	.519	-.2720	.520

BAND NORMALIZED EQUATIONS

BAND NORMALIZED CHANNEL OFFSET $\alpha' \equiv \sum C'_I \bar{Q}_I \equiv B =$ CHANNEL BIAS

BAND NORMALIZED CHANNEL GAIN $\beta' \equiv \sum D'_I \bar{Q}_I \equiv G =$ CHANNEL GAIN

WHERE BAND NORMALIZED REGRESSION COEFFICIENTS

$$C'_I = C_I + R_{\text{MIN}} D_I$$

$$D'_I = (R_{\text{MAX}} - R_{\text{MIN}}) D_I$$

BAND NORMALIZED VIDEO VALUE, VB, IS:

$$VB = \frac{127}{\beta'} (V - \alpha')$$

WHERE V IS THE RAW DIGITAL VIDEO VALUE AND THIS IS THE DIMENSIONLESS ABSOLUTE CALIBRATION EQUATION

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BAND NORMALIZED LINEAR REGRESSION COEFFICIENTS
FOR MSS BAND 1

CHANNEL												
	D_1'	C_1'	D_2'	C_2'	D_3'	C_3'	D_4'	C_4'	D_5'	C_5'	D_6'	C_6'
1	.403	-.051	.352	-.021	.296	.008	.248	.034	-.652	.514	-.653	.515
2	.419	-.049	.361	-.020	.309	.008	.257	.034	-.671	.513	-.674	.514
3	.409	-.049	.352	-.019	.303	.007	.254	.032	-.657	.513	-.660	.515
4	.419	-.049	.362	-.019	.309	.008	.260	.033	-.674	.513	-.677	.515
5	.419	-.050	.366	-.022	.303	.009	.255	.035	-.670	.513	-.674	.515
6	.420	-.050	.360	-.019	.307	.008	.260	.033	-.671	.513	-.675	.515

MSS/PF ABSOLUTE RADIOMETRIC ACCURACY

CONCLUSIONS

- INTEGRATING SPHERE RADIANCE CORRECT TO 10%
RELATIVE TO NBS STANDARD
- ERRORS FROM USE OF NOMINAL BAND WIDTHS
(PRIOR TO RECALIBRATION BY GE)

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<u>BAND</u>	<u>RANGE OF RADIANCE ERROR</u>	<u>RADIANCE BIAS</u>
1	1.8 %	- 9 %
2	12.9 %	7 %
3	2.6 %	-12 %
4	19.5 %	28 %

MSS RADIOMETRIC PRE-PROCESSING PROCEDURES

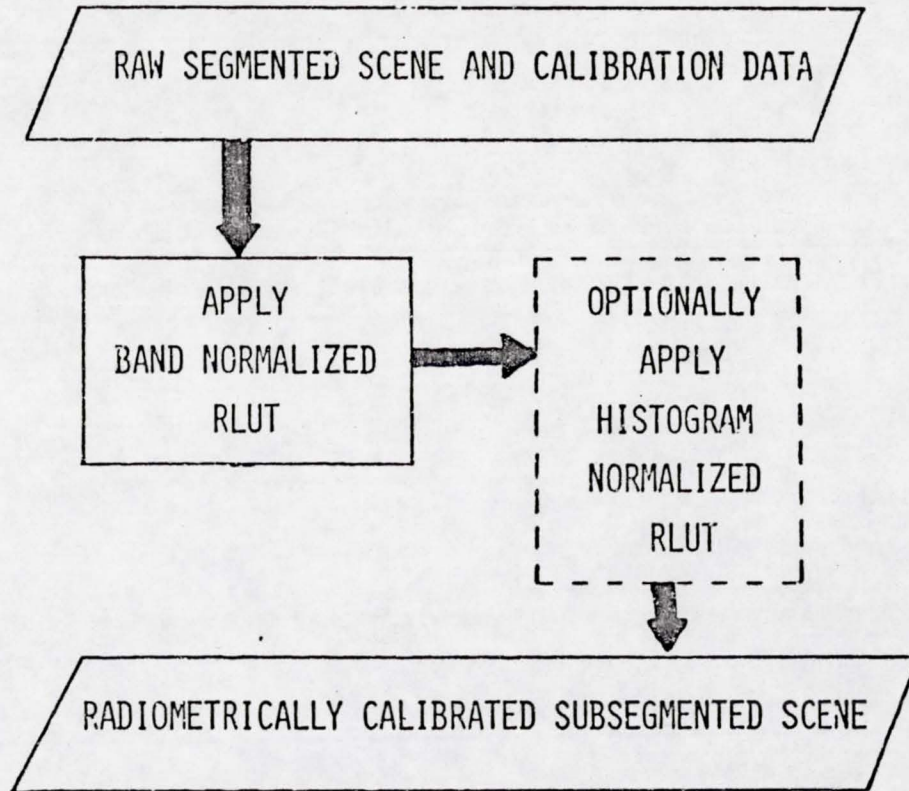
(MIPS = MSS IMAGE PROCESSING SYSTEM)

PURPOSE: CONVERT OBSERVED VIDEO DIGITAL COUNTS, V , INTO COUNTS WHICH ARE PROPORTIONAL TO RADIANCE, V_A , BY USING RADIOMETRIC LOOK-UP TABLES (RLUTs)

- STEPS:
- o COLLECT SCENE SEGMENT DATA
 - o CALCULATE BAND NORMALIZED GAIN, BIAS, AND RESULTING RLUT BY SEGMENT USING INTERNAL CALIBRATION LAMP DATA
 - o MODIFY GAIN AND BIAS USING SCENE HISTOGRAMS (OPTIONAL)
 - o GENERATE RLUT FOR EACH SUBSEGMENT BY BLENDING SEGMENT-LEVEL GAIN AND BIAS

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RADIOMETRIC PRE-PROCESSING DATA FLOW

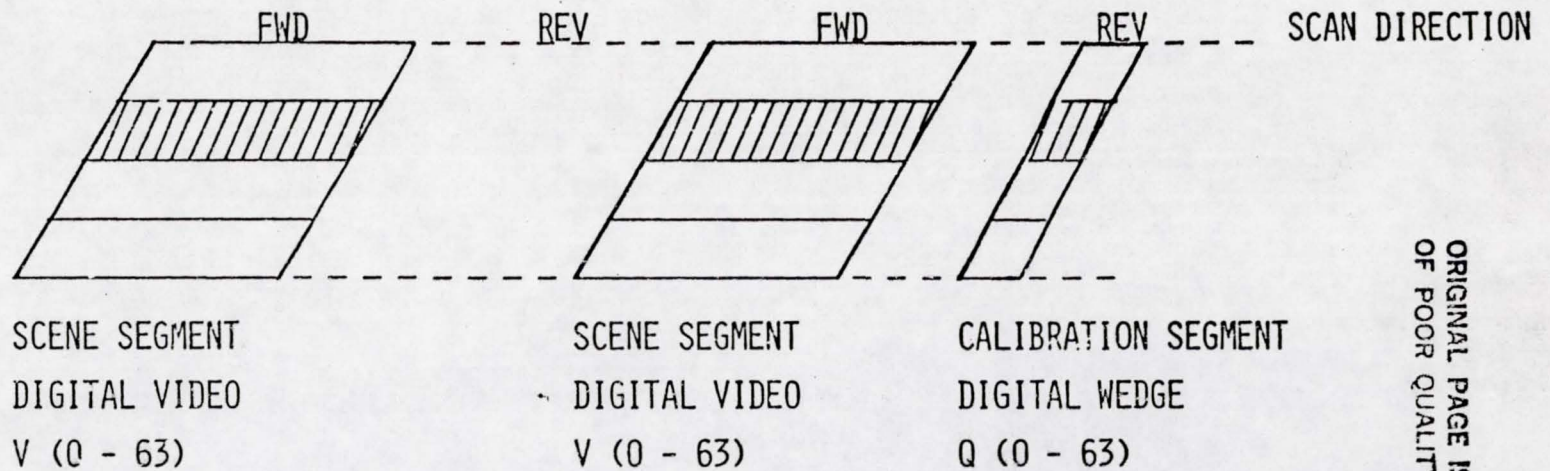


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COLLECT MSS SCENE SEGMENT DATA

DURING INITIAL INGEST OF RAW DATA INTO MIPS:

DIVIDE EACH SCENE INTO SEGMENTS (CAN BE ONE, TWO, FOUR, OR EIGHT)



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COLLECT SCENE HISTOGRAMS OF DIGITAL VIDEO VALUES, V, FOR EACH SEGMENT.

COLLECT SIX DIGITAL CALIBRATION WEDGE WORDS, Q, FROM EVERY OTHER LINE FOR EACH SEGMENT.

CALCULATE 24 BAND NORMALIZED RLUTs

FOR EACH SEGMENT FROM SELECTED CALIBRATION WEDGE DATA

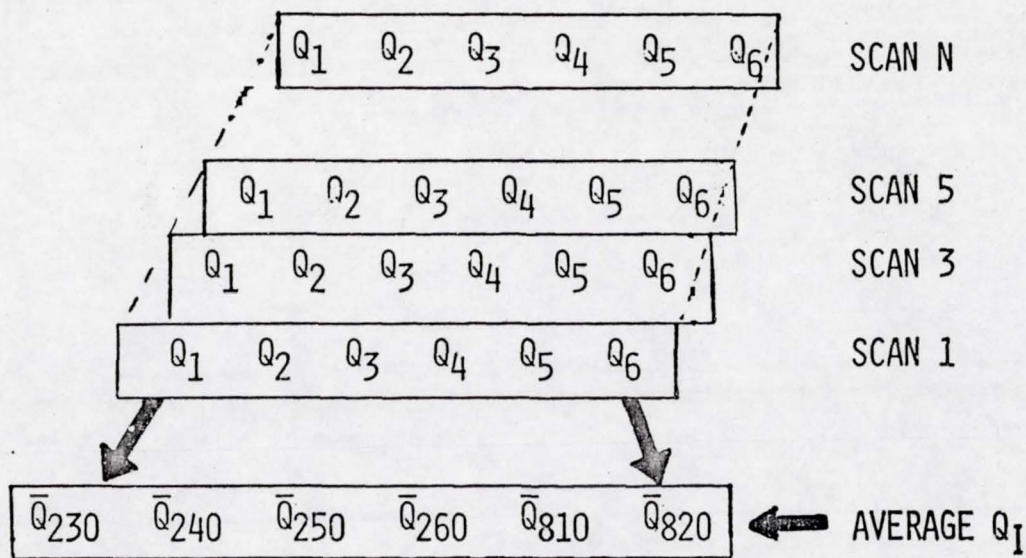
- 0 CALCULATE SIX AVERAGE CALIBRATION WEDGE VALUES, \bar{Q} , FOR EACH CHANNEL
- 0 CALCULATE INITIAL GAIN (G) AND BIAS (B) FOR EACH CHANNEL
- 0 CALCULATE POST-LAUNCH MODIFIERS
- 0 CALCULATE A BAND NORMALIZED RLUT BY CHANNEL

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CALCULATE SIX AVERAGE CALIBRATION WEDGE VALUES
FOR EACH CHANNEL IN THE SEGMENT

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- o DECOMPRESS THE SIX CALIBRATION WEDGE DIGITAL VALUES, Q , ACQUIRED ON EVERY OTHER SCAN, FROM (0 - 63) TO (0 - 127)
- o CALCULATE AVERAGE CALIBRATION WEDGE VALUE, \bar{Q} , FOR EACH WORD



WORD NUMBER I	WORD LOCATION N_I
1	230
2	240
3	250
4	260
5	810
6	820

CALCULATE INITIAL GAIN (G) AND BIAS (B)

FOR EACH CHANNEL IN THE SEGMENT

RETRIEVE ABSOLUTE LAMP CALIBRATION LINEAR REGRESSION COEFFICIENTS, C' AND D'
FROM DATA BASE :

CALCULATE GAIN AND BIAS FOR EACH CHANNEL:

$$G = \sum C' \bar{Q}$$

$$B = \sum D' \bar{Q}$$

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CALCULATE POST-LAUNCH MODIFIERS
(AS NECESSARY)

BASED ON HISTOGRAMMING OF REAL SCENE DATA,

CALCULATE MODIFIER (M) AND ADDER (A) TERMS TO CORRECT FOR RESIDUAL
STRIPING AND ATMOSPHERIC EFFECTS BY UPDATING THE ABSOLUTE CALIBRATION
EQUATION TO:

$$VC = \frac{127}{MG} (V-B) - A \equiv G'V + B'$$

M = 1 AND A = 0 UNTIL TIME OF UPDATE

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CALCULATE A BAND NORMALIZED RLUT
(FOR EACH CHANNEL)

COMPUTE A SEGMENT SPECIFIC RLUT:

$$\text{RLUT} = \text{INTEGER} [(i - 1) - B] / G$$

MAP RLUT INTO COMMON RANGE FROM 0 TO 127 TO AVOID STRIPING FROM
DETECTORS WITH DIFFERENT SENSITIVITY

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MODIFY BAND NORMALIZED GAIN AND BIAS USING SCENE HISTOGRAMS

FOR EACH CHANNEL IN THE SEGMENT

- o DECOMPRESS THE RAW SCENE HISTOGRAMS FROM (0 - 63) TO (0 - 127) (CONVERTS V-VALUES TO VD-VALUES IN ORDER TO CORRECT FOR NON-LINEARITY OF PHOTOMULTIPLIER TUBES)

- o CREATE CALIBRATED SCENE HISTOGRAMS BY APPLYING BAND NORMALIZED RLUT TO DECOMPRESSED SCENE HISTOGRAMS

- o CREATE A BAND AVERAGE SCENE HISTOGRAM FROM THE SIX INDIVIDUAL HISTOGRAMS

- o MODIFY EACH CHANNEL HISTOGRAM, R_H , SO THAT IT HAS THE SAME MEAN, $MEAN(R_H)$, AND STANDARD DEVIATION, $SD(R_H)$, AS THE BAND AVERAGE HISTOGRAM, $\overline{R_H}$, USING THE FOLLOWING FORMULA

HISTOGRAM NORMALIZATION (CONTINUED)

$$\overline{RH} = G * RH + b$$

$$g = SD(\overline{RH}) / SD(RH)$$

$$b = MEAN(\overline{RH}) - g * MEAN(RH)$$

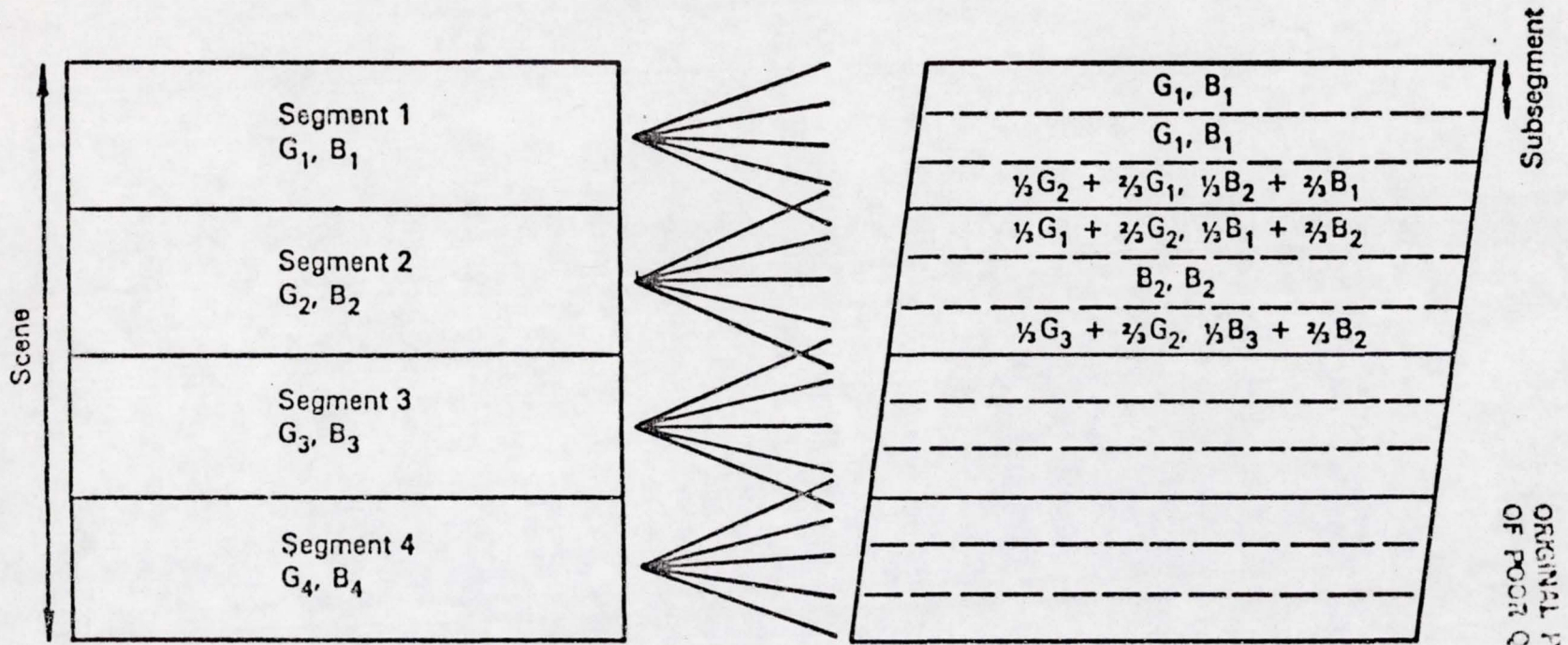
CALCULATE A HISTOGRAM NORMALIZED GAIN, G'' , AND BIAS, B''

$$G'' = G' / g$$

$$B'' = B - G' * b$$

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Scene Segment Blending



Example of Segment Blending With Number
Of Segments = 4
Number of Sub-Segments/Segment = 3

MSS Geometric Sensor Performance

Gary Banks

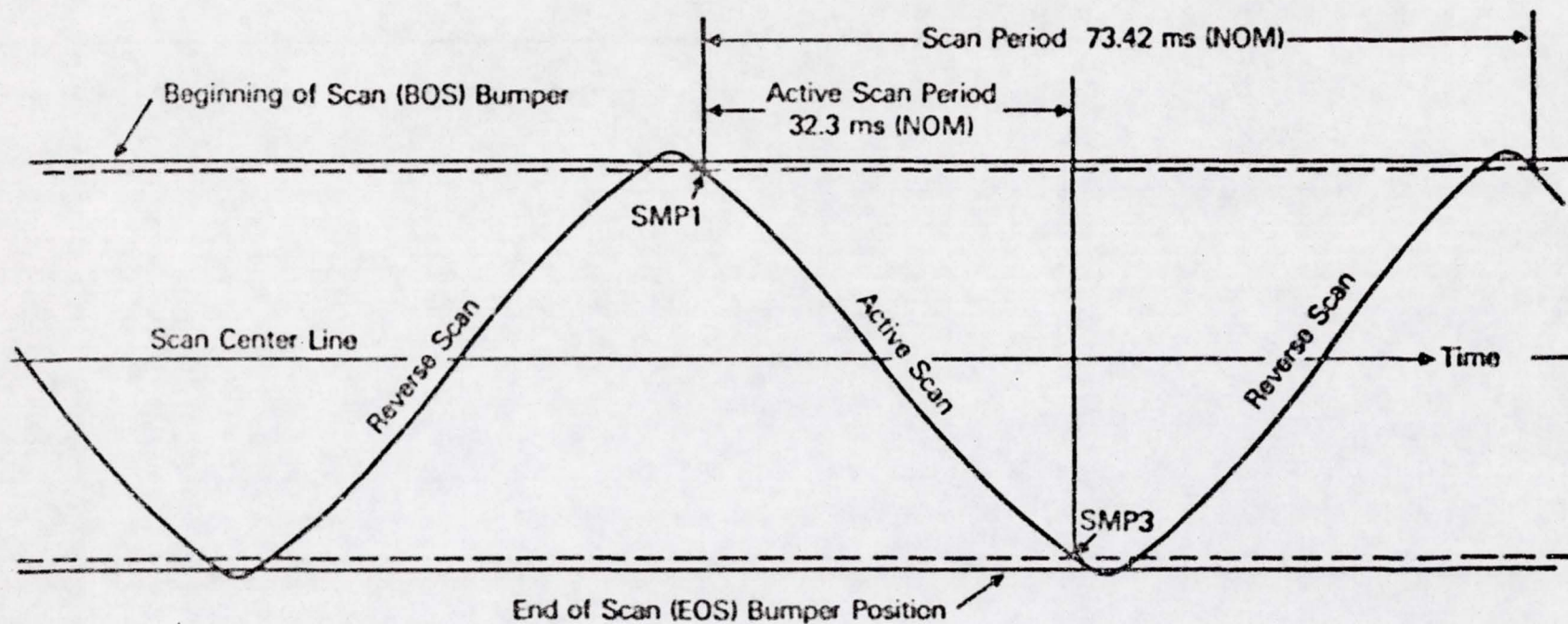
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Protoflight MSS-D Geometric Performance Summary

	SPEC	ACTUAL
Line Length Variation		
TM off	42 μ rad (rms)	12-19 μ rad (rms)
TM on	42 μ rad (rms)	109-113 μ rad (rms)
Line Length (Average)	31.5-34 ms	32.3 ms
Total Scan Angle	.26 \pm .001 rad	.2603 rad
Scan Repeatability		
TM off	24 μ rad (rms)	< 3 μ rad (rms)
TM on	24 μ rad (rms)	< 7 μ rad (rms)
Cross Scan		
Systematic	\pm 200 μ rad	< \pm 42 μ rad
Random	24 μ rad (1 σ)	< 3 μ rad
MTF (10 2 μ rad bars)		
Band 1	> .36	.49-.54
Band 2	> .36	.47-.54
Band 3	> .36	.47-.52
Band 4	> .36	.45-.48

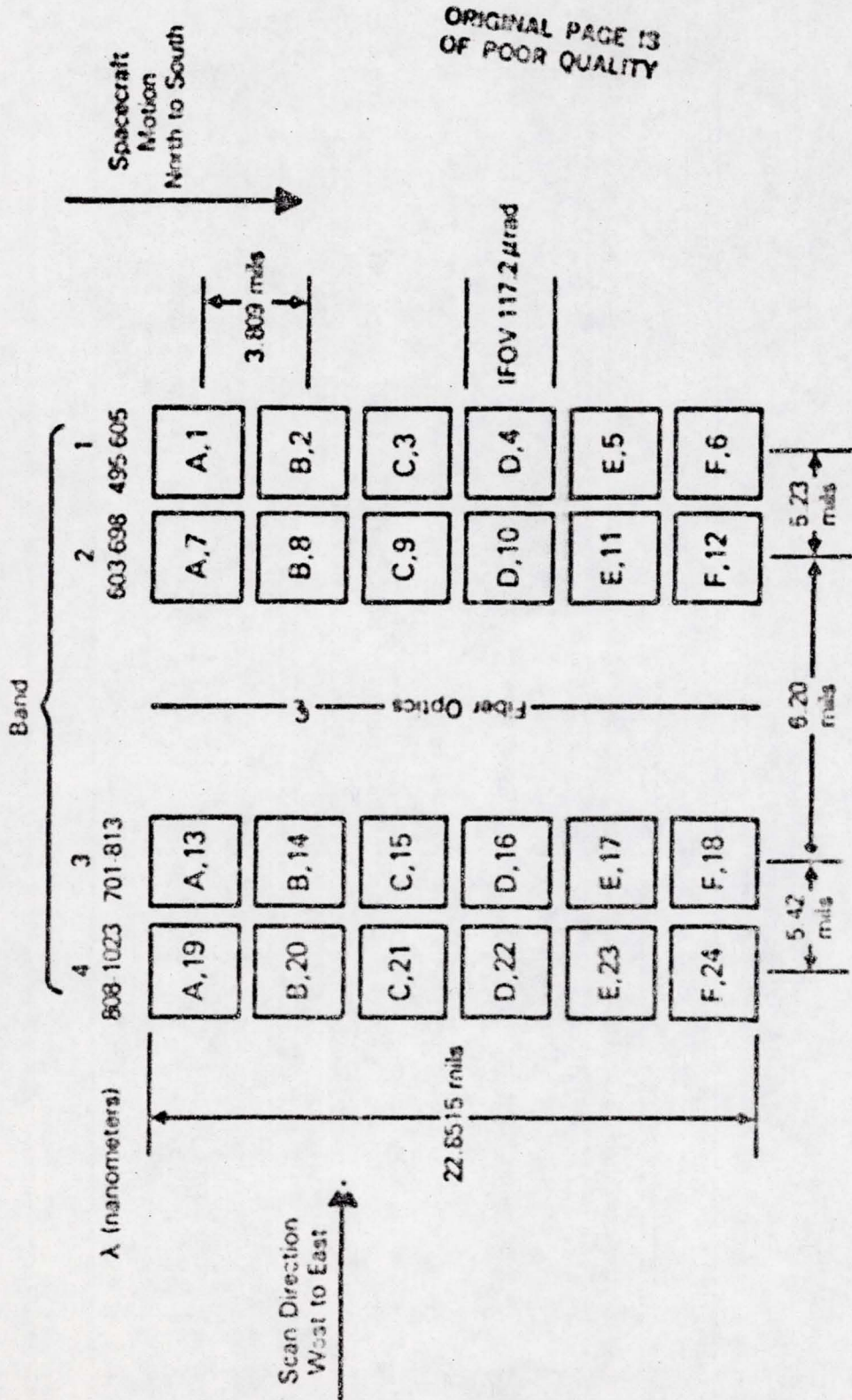
Scan Mirror Angle vs. Time Trajectory



SMP = Scan Monitor Pulse

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Landsat-D MSS/PF Focal Plane Dimensions



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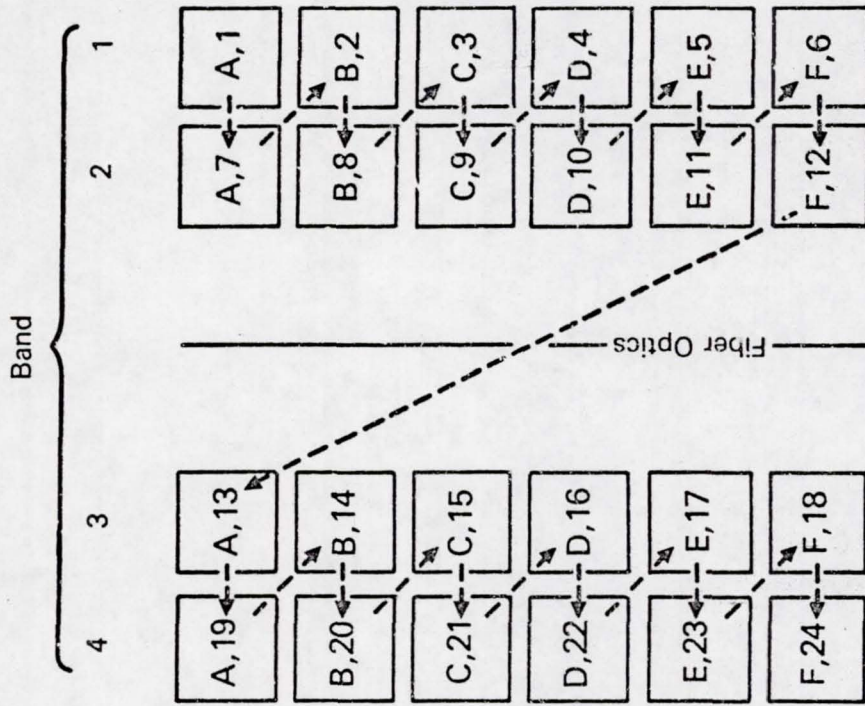
Note: Telescope Focal Length = 32.289 inches

1-194

mils = 0.001 inch.

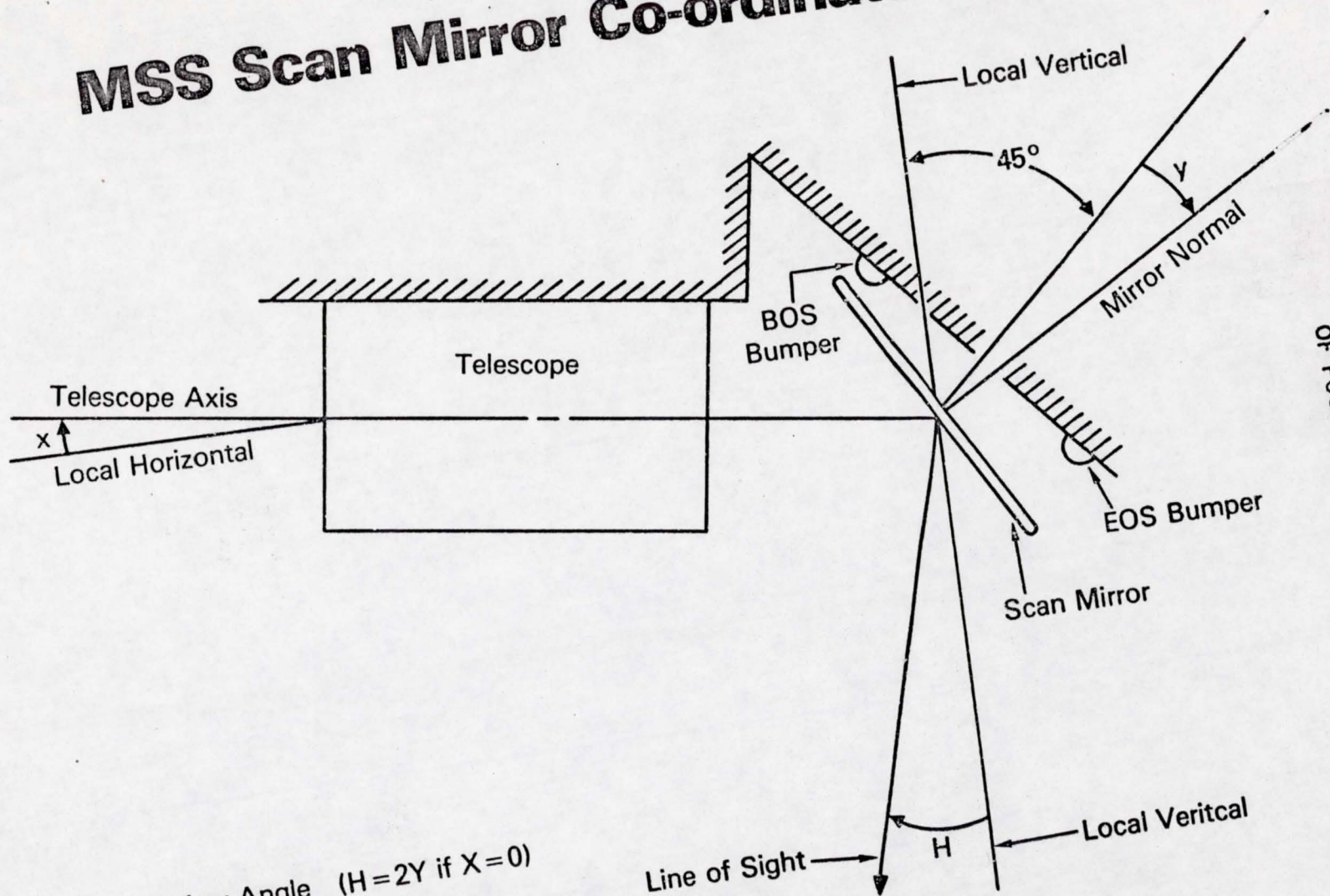
MSS Detector Sampling Sequence

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Sequence
Indicator

MSS Scan Mirror Co-ordinate System



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$H = \text{View Angle}$ ($H = 2Y$ if $X = 0$)
 $Y = \text{Mirror Angle}$
 $X = \text{Jitter Angle}$

Scan Profile Equations

$$I\ddot{y} + v\dot{y} + k(y - x) = 0$$

$$y = Ae^{-Rt} \sin \omega (t + F) + B \sin (\omega_0 t - \phi) + C \cos (\omega_0 t + \phi)$$

$$H = 2y - x$$

- Where:
- I = Mirror Moment of Inertia
 - y = Mirror Angle
 - v = Damping Coefficient
 - K = Flex Pivot Spring Constant
 - x = Jitter Angle = $A_0 \sin (\omega_0 t + \phi)$
 - H = View Angle
 - t = Time from Beginning of Scan (BOS)

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SCAN PROFILE EQUATIONS DEFINITION

$$A = \frac{-S_0 - A_0 \sin \phi + B \sin \phi + C \cos \phi}{\sin \omega F}$$

$$R = \frac{\omega_1^2}{2q}$$

$$F = \frac{1}{\omega} \tan^{-1} \left[\frac{\sin \omega P}{\frac{x(P) - y_p(P) - S_0}{y_p(0) - x(0) - S_0} e^{RP} + \cos \omega P} \right]$$

$$x(P) = A_0 \sin (\omega_0 P + \phi)$$

$$x(0) = A_0 \sin \phi$$

$$y_p(P) = B \sin (\omega_0 P + \phi) + C \cos (\omega_0 P + \phi)$$

$$y_p(0) = B \sin \phi + C \cos \phi$$

$$B = \frac{A_0 \left(1 - \frac{\omega_0^2}{\omega_1^2}\right)}{\left(1 - \frac{\omega_0^2}{\omega_1^2}\right)^2 + \left(\frac{\omega_0}{q}\right)^2 - A_0 \left(\frac{\omega_0}{q}\right)}$$

$$C = \frac{-A_0 \left(\frac{\omega_0}{q}\right)}{\left(1 - \frac{\omega_0^2}{\omega_1^2}\right)^2 + \left(\frac{\omega_0}{q}\right)^2}$$

$$\omega = \omega_1 \left[1 - \left(\frac{\omega_1}{2q}\right)^2\right]^{1/2}, \quad \omega_1 = \sqrt{\frac{k}{I}}, \quad q = \frac{k}{v}$$

Scan Profile

Zero Damping, Zero Jitter Case

$$x = v = 0$$

$$y = \frac{-S_0}{\sin \frac{\omega p}{2}} \sin (t - p/2)$$

$$H = 2y$$

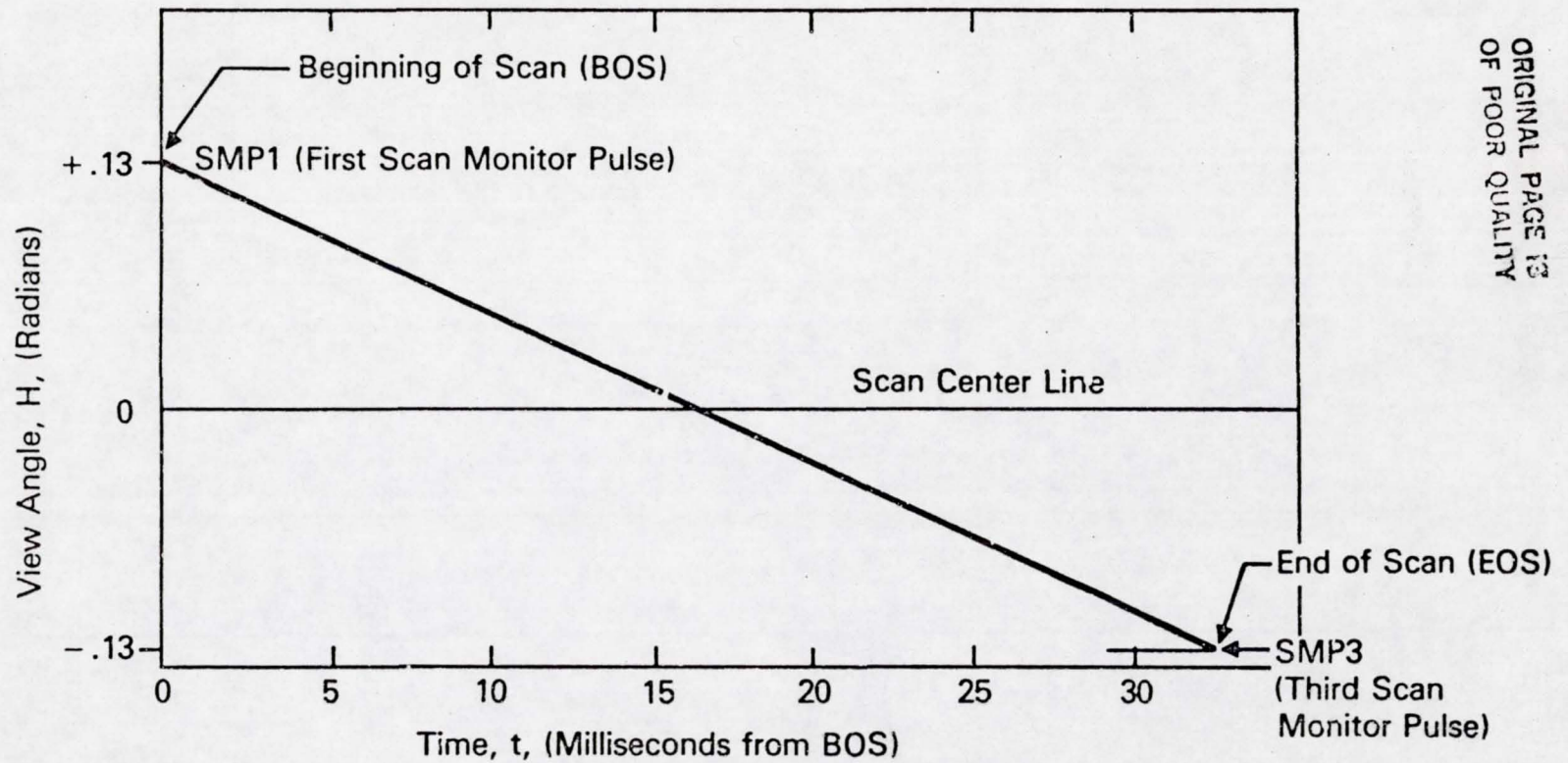
Where:

S_0 = Half of Mirror Scan Angle

p = Active Scan Period

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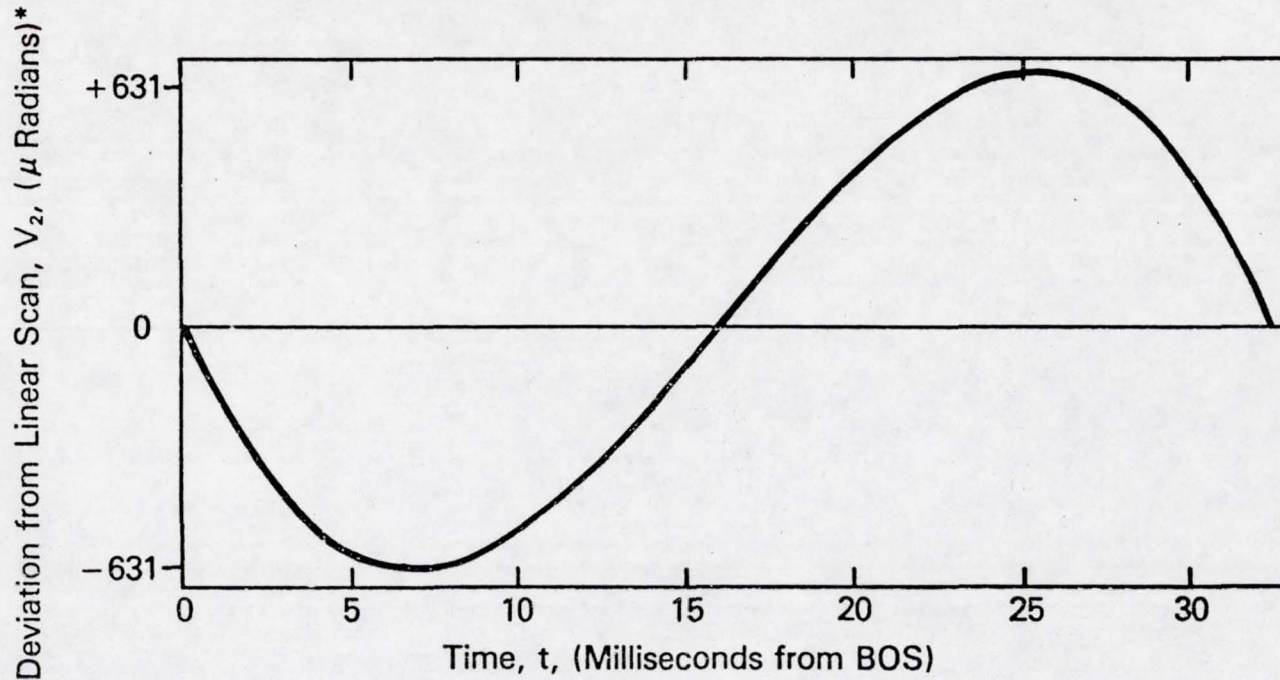
MSS Active Scan Mirror Profile



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Landsat-D MSS/PF Calculated Deviation from Linear Scan

(ASSUMES NO DAMPING)

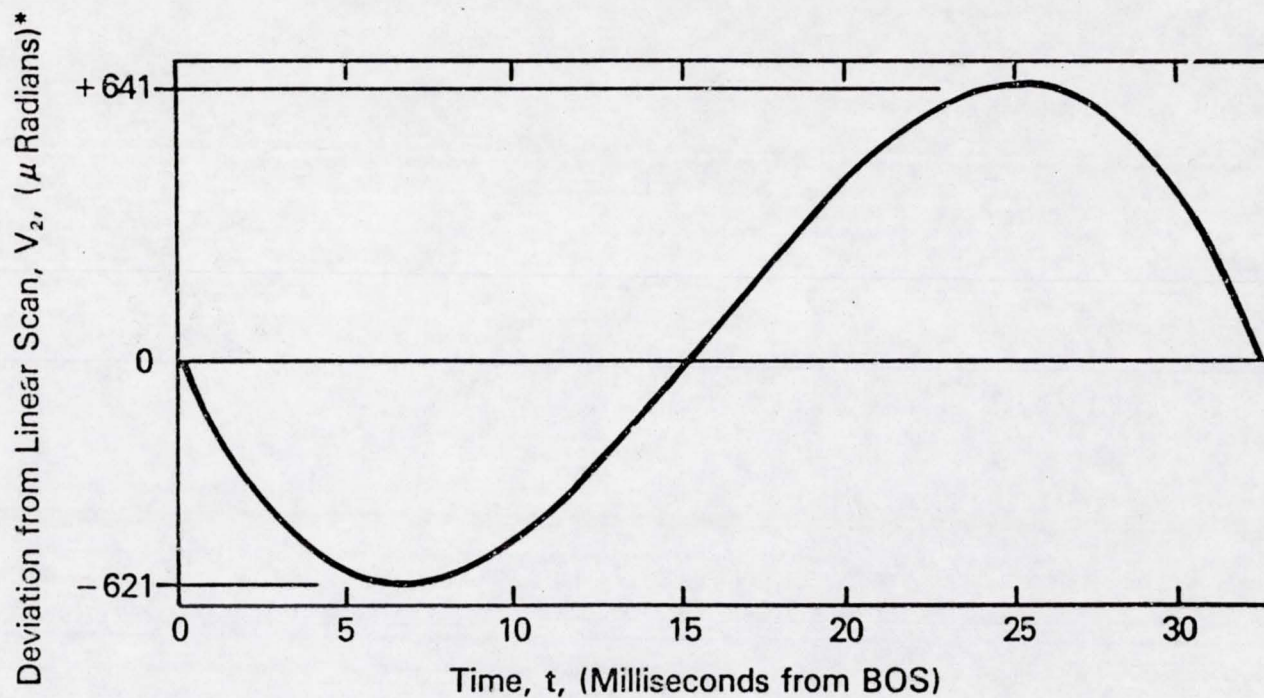


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*Nominal (Linear) Scan Profile Minus Actual Profile
(Scale to be Calibrated During System Level Tests)

Landsat-D MSS/PF Deviations from Linear Scan

(ASSUMES DAMPING, BASED ON
ENGINEERING MODEL MEASUREMENTS)



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*Scale to be Calibrated During System Level Tests

Protoflight MSS-D Scan Profiles
(with and without damping)

Definitions and physical values

- T - time in seconds from BOS (SMP-1)
- Y - mirror angle in radians
- H - view angle in radians
- P - active scan time (.03230 seconds)
- K - flex pivots spring constant (26.6 in lbs/rad)
- I - mirror inertia (.0923 in 16 sec²)
- S₀ - half of mirror angle from SMP-1 to SMP-3 (.065075 rad)

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Scan Profile with Damping Coefficient $\nu = 0$

T (SEC) = 0	H (RAD) = .13015000	Y (RAD) = 6.5075000E-02
T (SEC) = .0008075	H (RAD) = .12379434	Y (RAD) = 6.1897171E-02
T (SEC) = .001615	H (RAD) = .11741542	Y (RAD) = 5.8707711E-02
T (SEC) = .0024225	H (RAD) = .11101444	Y (RAD) = 5.5507218E-02
T (SEC) = .00323	H (RAD) = .10459259	Y (RAD) = 5.2296296E-02
T (SEC) = .0040375	H (RAD) = 9.8151091E-02	Y (RAD) = 4.9075546E-02
T (SEC) = .004845	H (RAD) = 9.1691147E-02	Y (RAD) = 4.5845574E-02
T (SEC) = .0056525	H (RAD) = 8.5213973E-02	Y (RAD) = 4.2606987E-02
T (SEC) = .00646	H (RAD) = 7.8720786E-02	Y (RAD) = 3.9360393E-02
T (SEC) = .0072675	H (RAD) = 7.2212807E-02	Y (RAD) = 3.6106403E-02
T (SEC) = .008075	H (RAD) = 6.5691258E-02	Y (RAD) = 3.2845629E-02
T (SEC) = .0088825	H (RAD) = 5.9157364E-02	Y (RAD) = 2.9578682E-02
T (SEC) = .00969	H (RAD) = 5.2612354E-02	Y (RAD) = 2.6306177E-02
T (SEC) = .0104975	H (RAD) = 4.6057457E-02	Y (RAD) = 2.3028729E-02
T (SEC) = .011305	H (RAD) = 3.9493906E-02	Y (RAD) = 1.9746953E-02
T (SEC) = .0121125	H (RAD) = 3.2922933E-02	Y (RAD) = 1.6461466E-02
T (SEC) = .01292	H (RAD) = 2.6345773E-02	Y (RAD) = 1.3172807E-02
T (SEC) = .0137275	H (RAD) = 1.9763663E-02	Y (RAD) = 9.8818316E-03
T (SEC) = .014535	H (RAD) = 1.3177839E-02	Y (RAD) = 6.5889195E-03
T (SEC) = .0153425	H (RAD) = 6.5895386E-03	Y (RAD) = 3.2947693E-03
T (SEC) = .01615	H (RAD) = -1.5154374E-10	Y (RAD) = -7.5771868E-11
T (SEC) = .0169575	H (RAD) = -6.5895389E-03	Y (RAD) = -3.2947694E-03
T (SEC) = .017765	H (RAD) = -1.3177839E-02	Y (RAD) = -6.5889196E-03
T (SEC) = .0185725	H (RAD) = -1.9763663E-02	Y (RAD) = -9.8818317E-03
T (SEC) = .01938	H (RAD) = -2.6345774E-02	Y (RAD) = -1.3172807E-02

T (SEC) = .0201875	H (RAD) = -3.2922933E-02	Y (RAD) = -1.6461467E-02
T (SEC) = .020995	H (RAD) = -3.9493906E-02	Y (RAD) = -1.9746953E-02
T (SEC) = .0218025	H (RAD) = -4.6057458E-02	Y (RAD) = -2.3028729E-02
T (SEC) = .02261	H (RAD) = -5.2612354E-02	Y (RAD) = -2.6306177E-02
T (SEC) = .0234175	H (RAD) = -5.9157364E-02	Y (RAD) = -2.9578682E-02
T (SEC) = .024225	H (RAD) = -6.5691258E-02	Y (RAD) = -3.2845629E-02
T (SEC) = .0250325	H (RAD) = -7.2212807E-02	Y (RAD) = -3.6196404E-02
T (SEC) = .02584	H (RAD) = -7.8720787E-02	Y (RAD) = -3.9360393E-02
T (SEC) = .0266475	H (RAD) = -8.5213973E-02	Y (RAD) = -4.2606987E-02
T (SEC) = .027455	H (RAD) = -9.1691147E-02	Y (RAD) = -4.5845574E-02
T (SEC) = .0282625	H (RAD) = -9.8151091E-02	Y (RAD) = -4.9075546E-02
T (SEC) = .02907	H (RAD) = -1.0459259	Y (RAD) = -5.2296296E-02
T (SEC) = .0298775	H (RAD) = -1.1101444	Y (RAD) = -5.5507218E-02
T (SEC) = .030685	H (RAD) = -1.1741542	Y (RAD) = -5.8707711E-02
T (SEC) = .0314925	H (RAD) = -1.2379434	Y (RAD) = -6.1897171E-02
T (SEC) = .0323	H (RAD) = -1.3015000	Y (RAD) = -6.5075000E-02

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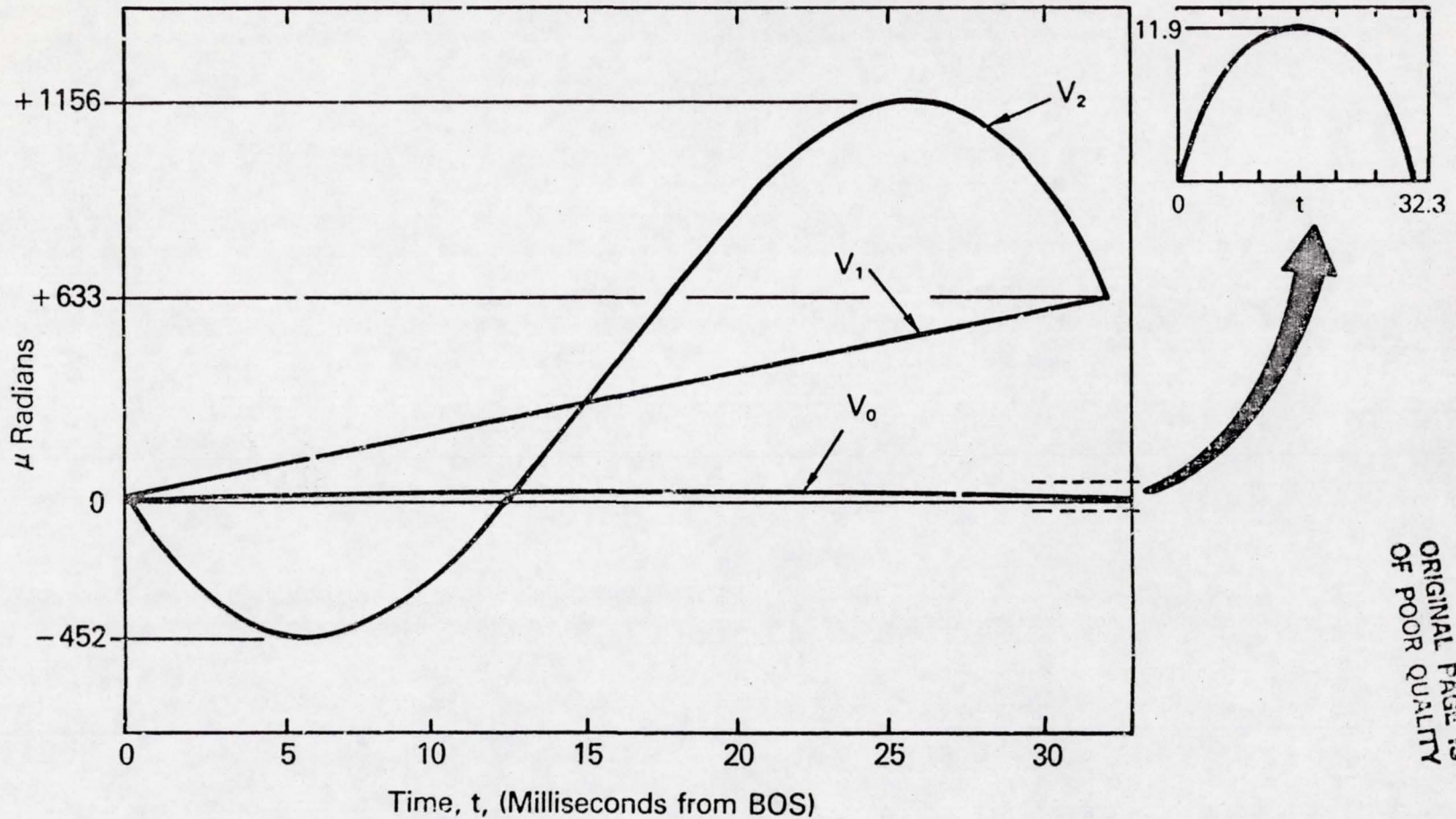
Scan Profile with Damping Coefficient $\nu = 107 \times 10^{-6}$ ft. - 16 - sec/rad

T (SEC) = 0	H (RAD) = .13015000	Y (RAD) = 6.5075000E-02
T (SEC) = .0008075	H (RAD) = .12379288	Y (RAD) = 6.1896439E-02
T (SEC) = .001613	H (RAD) = .11741256	Y (RAD) = 5.8706282E-02
T (SEC) = .0024225	H (RAD) = .11101026	Y (RAD) = 5.5505130E-02
T (SEC) = .00323	H (RAD) = .10458717	Y (RAD) = 5.2293583E-02
T (SEC) = .0040375	H (RAD) = 9.8144493E-02	Y (RAD) = 4.9072246E-02
T (SEC) = .004845	H (RAD) = 9.1683448E-02	Y (RAD) = 4.5841724E-02
T (SEC) = .0056525	H (RAD) = 8.5205248E-02	Y (RAD) = 4.2602624E-02
T (SEC) = .00646	H (RAD) = 7.8711110E-02	Y (RAD) = 3.9355553E-02
T (SEC) = .0072675	H (RAD) = 7.2202253E-02	Y (RAD) = 3.6101127E-02
T (SEC) = .008075	H (RAD) = 6.5679902E-02	Y (RAD) = 3.2839951E-02
T (SEC) = .0088825	H (RAD) = 5.9145282E-02	Y (RAD) = 2.9572641E-02
T (SEC) = .00969	H (RAD) = 5.2599622E-02	Y (RAD) = 2.6299811E-02
T (SEC) = .0104975	H (RAD) = 4.6044150E-02	Y (RAD) = 2.3022075E-02
T (SEC) = .011305	H (RAD) = 3.9480101E-02	Y (RAD) = 1.9740050E-02
T (SEC) = .0121125	H (RAD) = 3.2908705E-02	Y (RAD) = 1.6454353E-02
T (SEC) = .01292	H (RAD) = 2.6331201E-02	Y (RAD) = 1.3165600E-02
T (SEC) = .0137275	H (RAD) = 1.9748822E-02	Y (RAD) = 9.8744108E-03
T (SEC) = .014535	H (RAD) = 1.3162805E-02	Y (RAD) = 6.5814027E-03
T (SEC) = .0153425	H (RAD) = 6.5743898E-03	Y (RAD) = 3.2871949E-03
T (SEC) = .01615	H (RAD) = -1.5187312E-03	Y (RAD) = -7.5936560E-06
T (SEC) = .0169575	H (RAD) = -6.6046875E-03	Y (RAD) = -3.3023438E-03
T (SEC) = .017765	H (RAD) = -1.3192873E-02	Y (RAD) = -6.5964363E-03
T (SEC) = .0185725	H (RAD) = -1.9778505E-02	Y (RAD) = -9.8892523E-03
T (SEC) = .01938	H (RAD) = -2.6360346E-02	Y (RAD) = -1.3180173E-02
T (SEC) = .0201875	H (RAD) = -3.2937160E-02	Y (RAD) = -1.6469580E-02

T (SEC) = .020995	H (RAD) = -3.9507711E-02	Y (RAD) = -1.9753855E-02
T (SEC) = .0218025	H (RAD) = -4.6070764E-02	Y (RAD) = -2.3035382E-02
T (SEC) = .02261	H (RAD) = -5.2625086E-02	Y (RAD) = -2.6312543E-02
T (SEC) = .0234175	H (RAD) = -5.9169445E-02	Y (RAD) = -2.9584723E-02
T (SEC) = .024225	H (RAD) = -6.5702617E-02	Y (RAD) = -3.2851306E-02
T (SEC) = .0250325	H (RAD) = -7.223360E-02	Y (RAD) = -3.6111680E-02
T (SEC) = .02584	H (RAD) = -7.8730463E-02	Y (RAD) = -3.9365231E-02
T (SEC) = .0266475	H (RAD) = -8.5222698E-02	Y (RAD) = -4.2611345E-02
T (SEC) = .027455	H (RAD) = -9.1698845E-02	Y (RAD) = -4.5849423E-02
T (SEC) = .0282625	H (RAD) = -9.8157689E-02	Y (RAD) = -4.9078845E-02
T (SEC) = .02907	H (RAD) = -1.0459802	Y (RAD) = -5.2295008E-02
T (SEC) = .0298775	H (RAD) = -1.1101861	Y (RAD) = -5.5509307E-02
T (SEC) = .030685	H (RAD) = -1.1741928	Y (RAD) = -5.8709139E-02
T (SEC) = .0314925	H (RAD) = -1.2379581	Y (RAD) = -6.1897903E-02
T (SEC) = .0323	H (RAD) = -1.3015000	Y (RAD) = -6.5075000E-02

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Line Length Variation Effects on Scan Mirror Profile



Line Length Error = $-78 \mu\text{sec}$.

V_0 = Error Remaining after "Rubberband" Correction

V_1 = Error Resulting from Shortened Line Length if Baseline Profile is Used

V_2 = Deviation from Linear Scan

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MSS Geometric Processing and Calibration

Joan Brooks

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GEOMETRIC CORRECTION AGENDA

- REQUIREMENTS
- OVERVIEW OF CORRECTION DATA GENERATION
- SYSTEMATIC CORRECTION DATA GENERATION
- GEODETIC CORRECTION DATA GENERATION
- GEOMETRIC CORRECTION CALIBRATION

SUMMARY OF KEY GEOMETRIC CORRECTION REQUIREMENTS

- **COMPLETE GEOMETRIC CORRECTION CALCULATION**
 - EXCEPT NO TERRAIN RELIEF COMPENSATION
- **USE EPHEMERIS FROM GPS OR ORBIT SUPPORT COMPUTING DIVISION (2 DAY PREDICT)**
- **MAP PROJECTIONS**
 - UNIVERSAL TRANSVERSE MERCATOR/POLAR STEREOGRAPHIC
 - SPACE OBLIQUE MERCATOR
- **INTERACTIVE CONTROL POINT LIBRARY BUILD**
 - SELECT CONTROL POINTS FROM MAPS AND PHOTOGRAPHIC IMAGERY
 - 100 CONTROL POINTS/DAY
 - CAPABILITY TO USE THE EXISTING LANDSAT 2/3 LIBRARY
 - ELEVATION OF CONTROL POINTS MUST BE USED
- **SCENE FRAMING BASED UPON A WORLD REFERENCE SYSTEM (WRS)**
- **FULLY CORRECTED MSS IMAGERY**
 - FOR SENSOR AND PROCESS QUALITY ASSESSMENT
 - RESAMPLING USING NEAREST NEIGHBOR OR CUBIC CONVOLUTION
 - CCT AND 241 MM FILM OUTPUTS
- **MSS ARCHIVAL HIGH DENSITY TAPE**
 - FORMAT AND CONTENT MUST CONFORM WITH IPF-ICD-201
 - DEFINES MSS GEOMETRIC CORRECTION DATA

SUMMARY OF KEY GEOMETRIC CORRECTION REQUIREMENTS (ACCURACY)

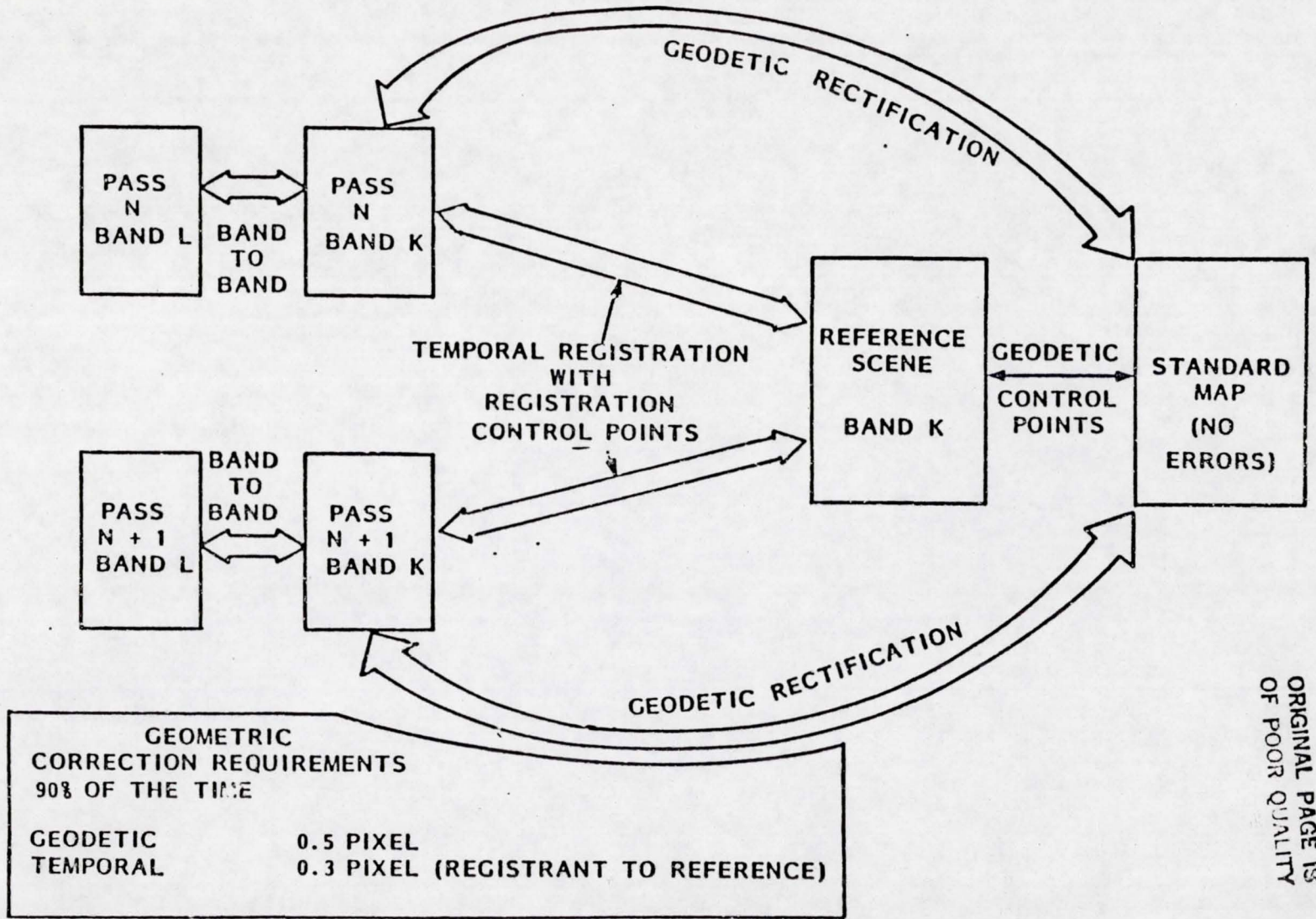
GEODETIC RECTIFICATION

- 0.5 PIXEL (90% OF THE TIME)
- REFERENCE TO STANDARD MAP
- ASSUME ACCURATE GROUND CONTROL POINTS
- VERIFIED OVER AREAS WITH NO TOPOGRAPHICAL VARIATIONS

TEMPORAL REGISTRATION

- 0.3 PIXEL (90% OF THE TIME)
- ADEQUATE INSTRUMENT PERFORMANCE

GEOMETRIC ACCURACY SPECIFICATIONS



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TEMPORAL REGISTRATION REQUIREMENT INTERPRETATION

- TEMPORAL REGISTRATION REQUIREMENT: 0.3 PIXEL, 90% OF THE TIME

- CLARIFICATIONS
 - SPECIFICATION APPLIES PER AXIS OF THE OUTPUT SCENE (X, Y)
 - PIXEL DEFINED AS INPUT PIXEL SIZE
 - 42.5 μ RADIAN FOR TM
 - 117.2 μ RADIAN FOR MSS
 - (AVOIDS ALTITUDE EFFECTS AND RESTRAINT OF REDUCED TM OUTPUT PIXEL SIZE, 28.5 METER)
 - ADEQUATE NUMBER AND DISTRIBUTION OF CONTROL POINTS
 - ELEVATION OF CONTROL POINTS MUST BE KNOWN
 - VERIFIED OVER AREAS WITH NEGLIGIBLE EARTH TOPOLOGICAL VARIATIONS

TEMPORAL REGISTRATION VERIFICATION

- CATEGORIES OF ERRORS

- BIASES: "FIXED" OFFSET BETWEEN REFERENCE AND SUBSEQUENT PASS. RESULTS PRIMARILY FROM ATTITUDE, ALIGNMENT AND EPHEMERIS UNCERTAINTY. RANDOM OVER THE ENSEMBLE OF ESTIMATION EVENTS. ENSEMBLE VARIANCE σ_B^2 .
- RANDOM: INTERNAL ERRORS WITHIN ONE SCENE. RESULTS FROM SCAN MIRROR NON-REPEATABILITY, RESIDUAL JITTER, PROCESSING LINEARIZATION AND COMPUTATIONAL LIMITATIONS. ONE SCENE VARIANCE σ_R^2 .
- MEASUREMENT: ERROR IN CORRELATING TWO CONTROL POINTS. VARIANCE σ_M^2 .

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MULTISPECTRAL SCANNER TEMPORAL REGISTRATION ERROR IN PIXEL (117.2 μ r) 90% OF THE TIME

ERROR SOURCE	MSS-D PERFORMING AT SPEC LEVEL				MSS-D PERFORMING AT MSS-3 LEVEL			
	CROSS TRACK		ALONG TRACK		CROSS TRACK		ALONG TRACK	
	BIAS	RANDOM	BIAS	RANDOM	BIAS	RANDOM	BIAS	RANDOM
<u>MULTISPECTRAL SCANNER</u>								
SCAN REPEATABILITY	-	.337 $\sqrt{2}$	-	.463 $\sqrt{2}$	-	.112 $\sqrt{2}$	-	.049 $\sqrt{2}$
BAND-TO-BAND	-	-	-	-	-	-	-	-
SCAN UNDERLAP/OVERLAP	-	-	-	.059 $\sqrt{2}$	-	-	-	.059 $\sqrt{2}$
WORST CASE 45°N	-	-	-	.126 $\sqrt{2}$	-	-	-	.126 $\sqrt{2}$
81.8°S	-	-	-	-	-	-	-	-
<u>SPACECRAFT</u>								
JITTER	-	.102 $\sqrt{2}$	-	.102 $\sqrt{2}$	-	.102 $\sqrt{2}$	-	.102 $\sqrt{2}$
<u>GROUND PROCESSING</u>								
• ATTITUDE/EPIHEMERIS RESIDUAL	.269	-	.318	-	.165	-	.165	-
• SYSTEMATIC CORRECTION DATA GENERATION	-	.030 $\sqrt{2}$	-	.030 $\sqrt{2}$	-	.030 $\sqrt{2}$	-	.030 $\sqrt{2}$
• GEOMETRIC CORRECTION DATA GENERATION	-	.030 $\sqrt{2}$	-	.030 $\sqrt{2}$	-	.030 $\sqrt{2}$	-	.030 $\sqrt{2}$
• CORRECTION DATA INTER- POLATION	-	.030 $\sqrt{2}$	-	.030 $\sqrt{2}$	-	.030 $\sqrt{2}$	-	.030 $\sqrt{2}$
• RESAMPLING	-	.014 $\sqrt{2}$	-	.014 $\sqrt{2}$	-	.014 $\sqrt{2}$	-	.014 $\sqrt{2}$
RSS SUBTOTAL	.269	.505	.318	.699	.165	.230	.165	.254 (81.8°S) .199 (45°N)
RANDOM + BIAS RSS	.572		.768 (81.8°S)		.283		.303 (81.8°S) .259 (45°N)	
SYSTEM SPECIFICATION	.3		.3		.3		.3	

OVERVIEW OF CORRECTION DATA GENERATION

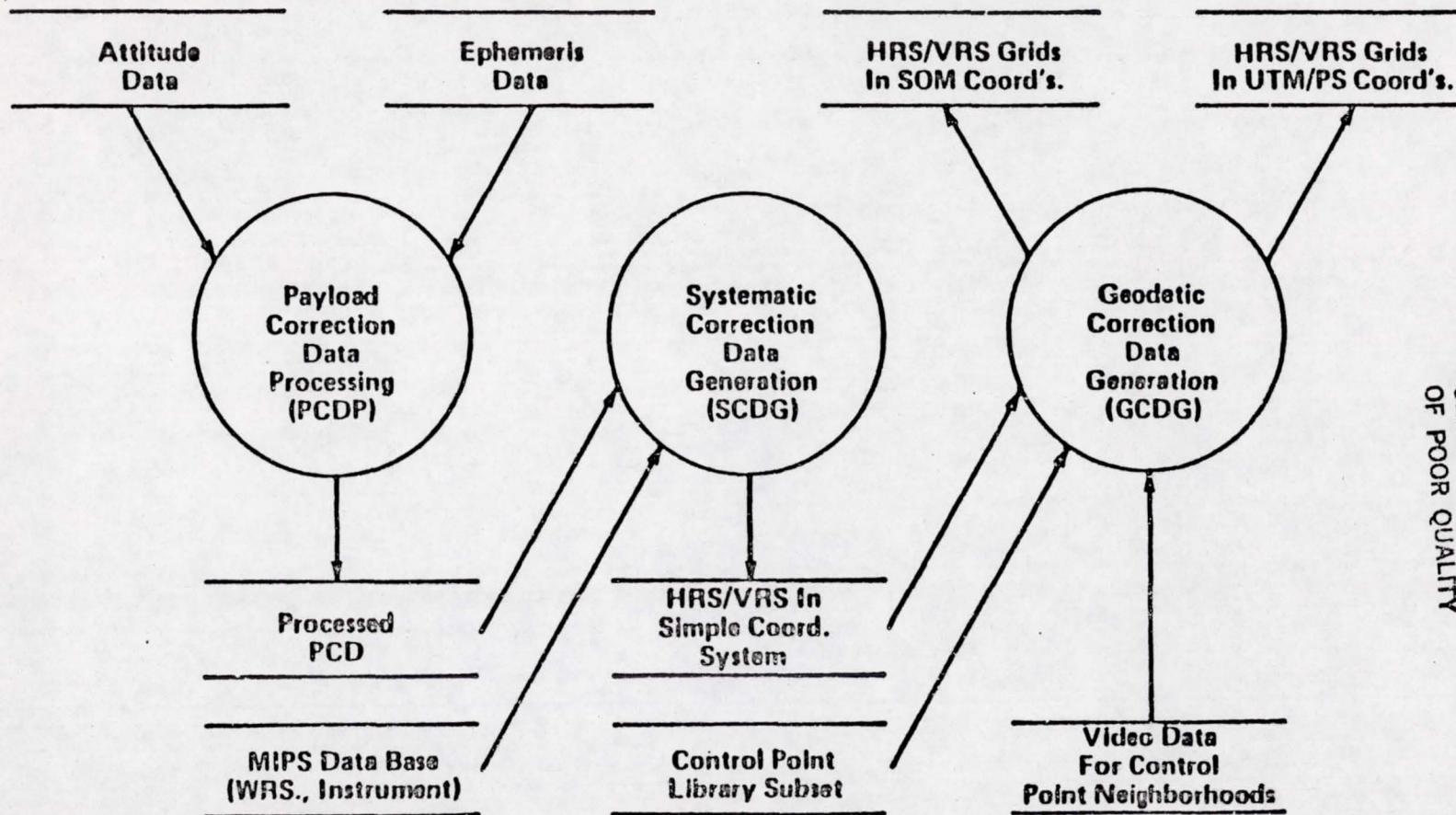
- SYSTEMATIC CORRECTION DATA - THREE STAGES
 - OFFLINE DEVELOPMENT OF NOMINAL CORRECTION DATA
 - ONLINE ATTITUDE/EPOCHERIS PROCESSING, SCENE CENTER DETERMINATION
 - ONLINE UPDATE OF NOMINAL CORRECTION DATA USING PROCESSED ATTITUDE/EPOCHERIS

- GEODETIC CORRECTION DATA - ONLINE
 - EXTRACTION AND CORRECTION OF CONTROL POINT NEIGHBORHOODS USING SYSTEMATIC CORRECTION DATA
 - CORRELATION OF CONTROL POINT LIBRARY CHIPS TO CONTROL POINT NEIGHBORHOODS
 - FILTERING OF CONTROL POINT DISLOCATION TO DETERMINE EPOCHERIS/ATTITUDE CORRECTIONS
 - UPDATE OF SYSTEMATIC CORRECTION DATA USING FILTER OUTPUTS

OVERVIEW OF GEOMETRIC CORRECTION (CONTINUED)

- REFORMATTING OF CORRECTION DATA TO HORIZONTAL AND VERTICAL RESAMPLING MATRICES (HRS/VRS)

Overview of MSS Geometric Correction Data Generation for a Scene



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Legend

HRS: Horizontal Resampling
VRS: Vertical Resampling
SOM: Space Oblique Mercator

UTM: Universal Transverse Mercator
PS: Polar Stereographic

Systematic Correction of Imagery: Effects Modeled

- **Spacecraft Ephemeris (Input)**
- **Spacecraft Attitude (Input)**
- **Scanner Misalignment (Parameters)**
- **Scan Angle Profile (Parameterized Model)**
- **Earth Geoid (Parameterized Model)**
- **Earth Rotation (Parameterized Model)**
- **World Reference System Scene Centers**
- **Calculations Performed for Single Detector in Center of
Detector Array —**
 - **Band-Line Adjustments (BLA) Applied Separately**
 - **Line Length Corrections Applied Separately**

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MSS Mirror Model for SCDG

$$\bullet \rho(i) = z \cdot e^{-\beta(t-t_0)} \{ A \sin \omega (t_0 + [i-1] \Delta t) \}$$

ρ = Scan Angle

i = Pixel Number

Δt = Sampling Time

t_0 = Line Start Time

ω = Mirror Frequency

A = Amplitude

β = DAMPING CONSTANT

- Along Scan Only (Roll Axis)
- Need Shape for Cross Scan (Pitch/Yaw vs i)

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Coordinate System For Basic Calculations

Standard Coordinate System:

- Origin at WRS Scene Center
- X-Axis Along S/C Nominal Angular Momentum Vector
- Z-Axis Earth Center Pointing
- Y-Axis Completes Right Hand Coordinate System (Parallel to SC Nominal Inertial Velocity)

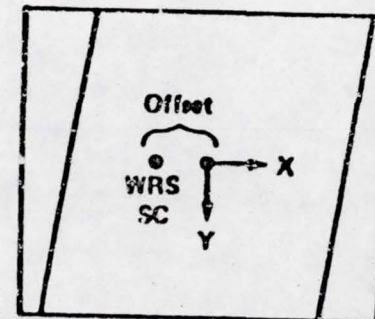
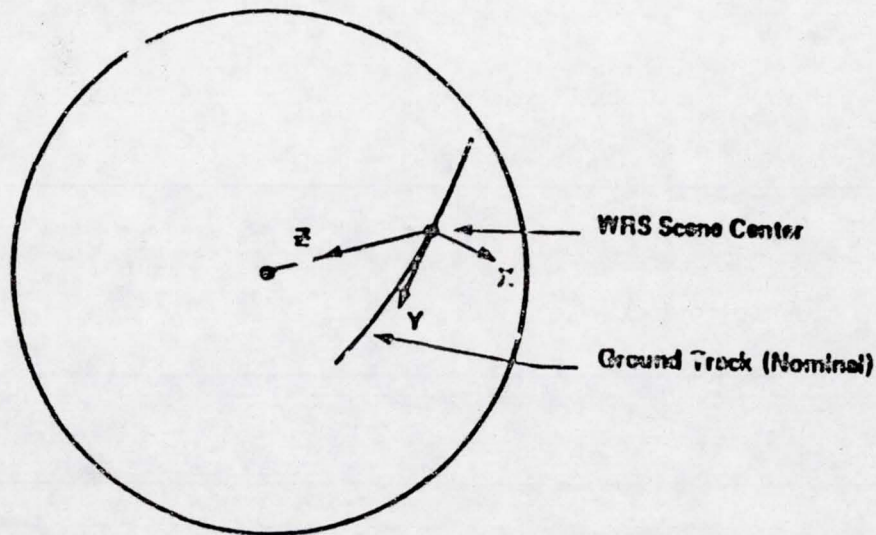
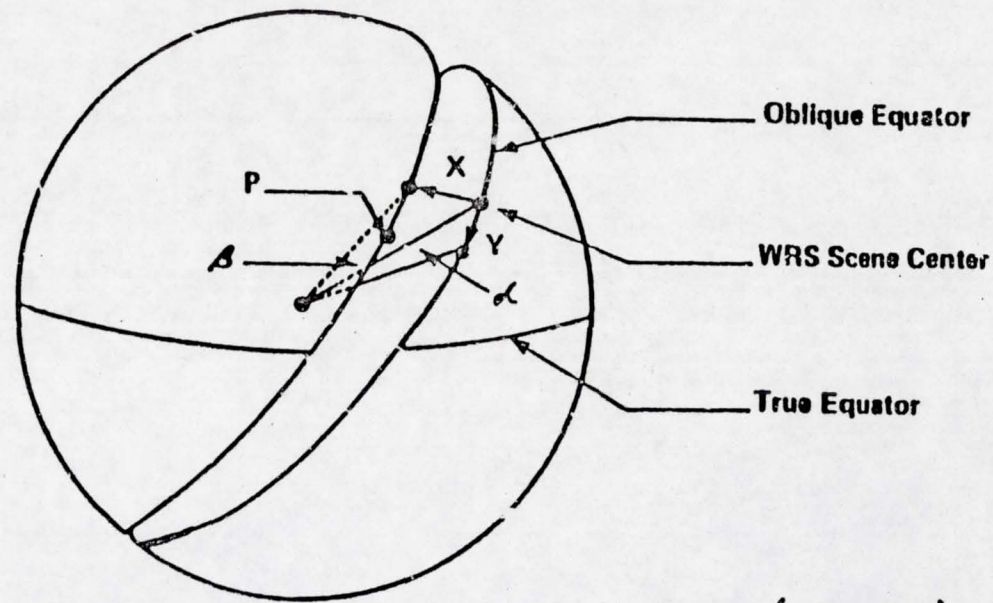


Image Centered
in Frame

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Oblique Mercator Coordinates For The Sphere



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L = Oblique Longitude of P
 B = Oblique Latitude of P
 R_0 = Local Earth Radius
 at WRS Scene Cent

$$X = \frac{R_0}{2} \text{LN} \left(\frac{1 + \text{SIN } B}{1 - \text{SIN } B} \right)$$

$$Y = R_0 \cdot L$$

- Highly Conformal With ⁱⁿ Scene
- Good Local Approximation to Space Oblique Mercator

Systematic Correction Functions (SCF)

Objective: Associate (X, Y) in Processed Image with (i, j) in Raw Video

i = Pixel Number (Sample Time)

j = Line Number (Scan Time)

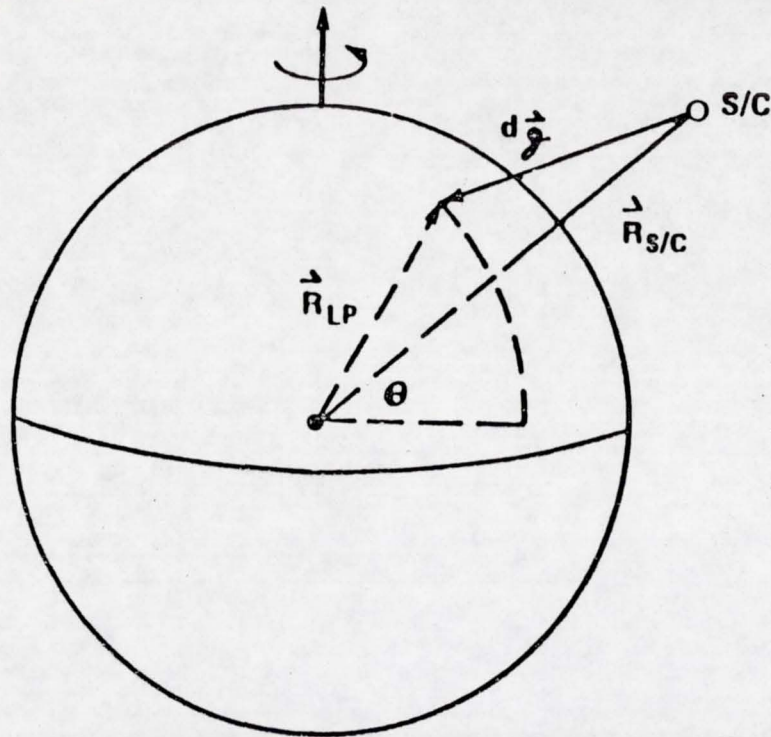
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(i, j) Defines Unique Time t Relative to Scene Center Time

**Result: $X = f(i, j)$
 $Y = g(i, j)$**

**For Given Ephemeris
and Attitude**

Systematic Corrections Functions: The Lookpoint Calculation



Ellipsoidal Earth

- In ECI Coordinates, solved for (θ, λ')

$$\vec{R}_{LP}(\theta, \lambda') = \vec{R}_{S/C}(t) + d\vec{g}(t)$$

t From Pixel Number i

Scan Line j

Scene Center Time

$\vec{R}_{S/C}(t)$ From Ephemeris

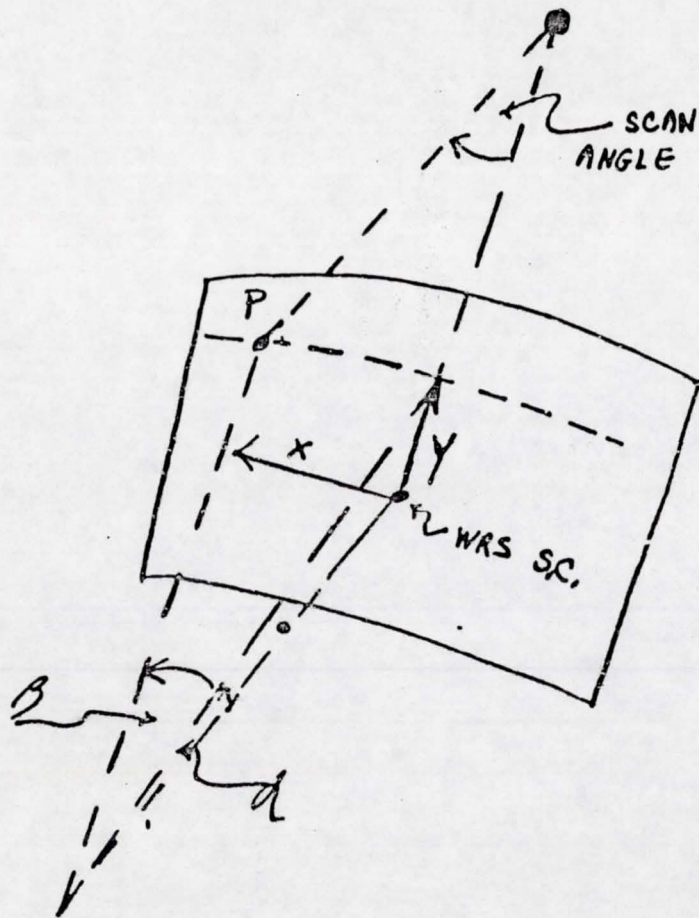
$\vec{g}(t)$ From Scan Profile,
Alignment, Attitude Data

- Transform to ECEF Coordinates
- $\lambda(i, j)$ From λ' , Earth Spin
- Transform (θ, λ) to (α, β)
- Transform (α, β) to (x, y)

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Nominal Systematic Correction Functions (SCF₀)

- Spacecraft in Orbit that Determines WRS Scene Centers (Nominal Orbit)
- Spacecraft and Sensor have Nominal Pointing (Nominal Attitude Data)



$$x_0 = f_0(i, j)$$

$$y_0 = g_0(i, j)$$

SCF₀

x_0, y_0 Independent
of Spacecraft
Ephemeris and Attitude

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Properties Of Systematic Correction Functions

- Highly Linear in Spacecraft Parameter Biases

$$x(i, f, \vec{\delta}) = x_0(i, f) + \sum_L \frac{\partial x(i, f)}{\partial \delta_L} \delta_L(i, f)$$

$$y(i, f, \vec{\delta}) = y_0(i, f) + \sum_L \frac{\partial y(i, f)}{\partial \delta_L} \delta_L(i, f)$$

- Analytic Expressions for Partial Derivatives Defined Over Entire Grid
- Spacecraft Parameter Biases

$\delta\alpha$ — Bias in Oblique Longitude
 $\delta\beta$ — Bias in Oblique Latitude
 δr — Bias in S/C Radial Location
 $\delta\theta_p$ — Pitch Angle
 $\delta\theta_x$ — Roll Angle
 $\delta\theta_y$ — Yaw Angle

- Biases Generally Functions of Time

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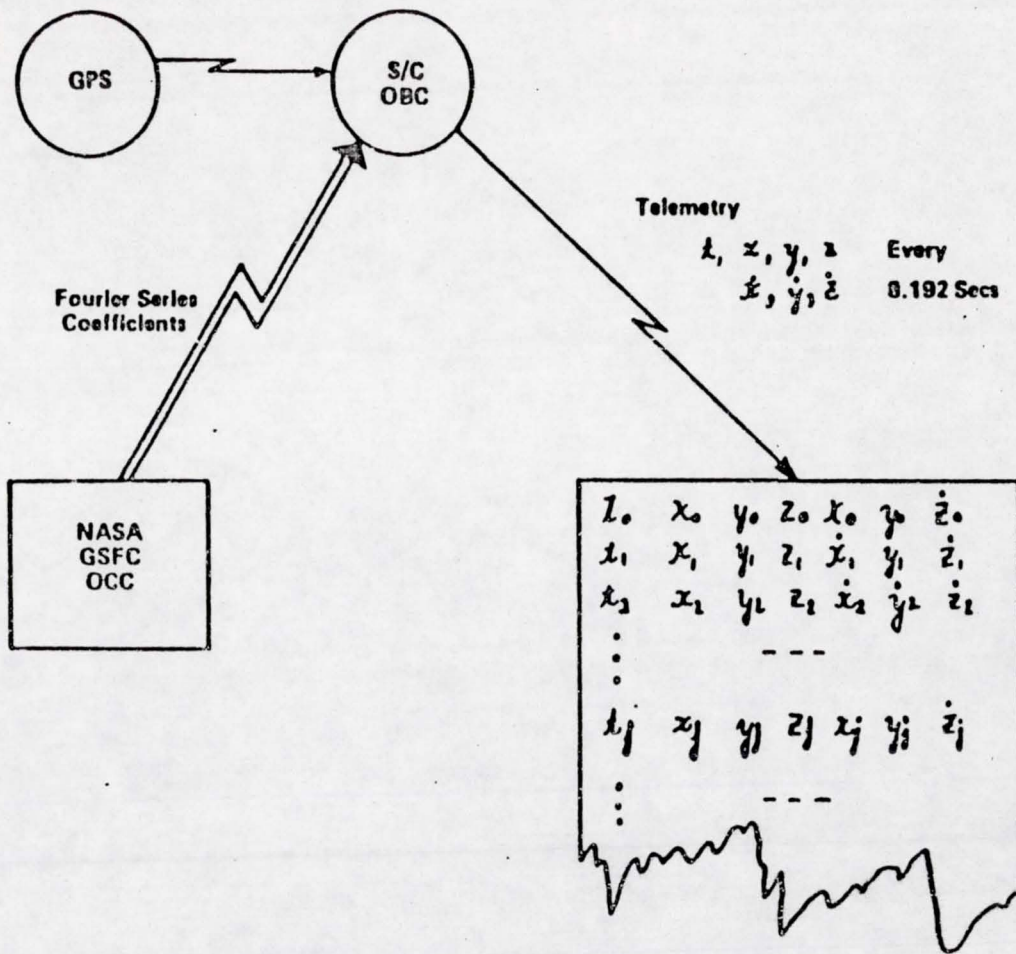
Offline Generation of Nominal Systematic Correction Functions/Partial Derivatives

- SCF₀ Generated on Grid 41 (Along Scan) x 11 (Along Track)
- Partial Derivatives (PD) Generated on Grid 5 (Along Scan) x 3 (Along Track)
- Linear Interpolation Suffices
- Functions of WRS Scene Center Latitude Only
- Stored for Single Reference Path, Along with Nominal Ephemeris and WRS Scene Center Data
- Total Data Base < 1.5 MBytes

Ephemeris Processing

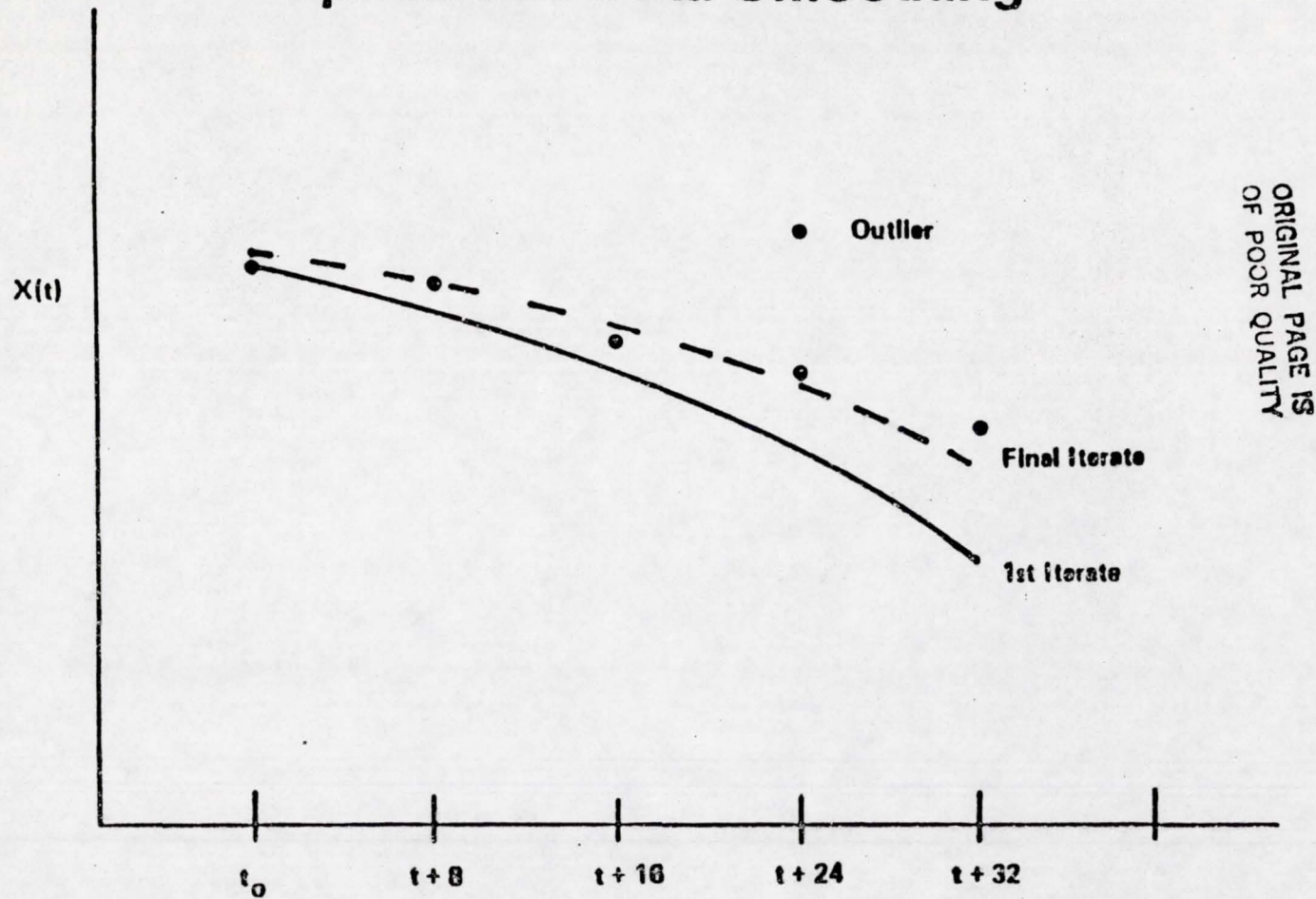
- **Received Ephemeris has been Processed on the OBC**
- **May Be Contaminated By**
 - **Noise**
 - **Outliers (Transmission Errors)**
 - **Bias (Ground Truth Required for Removal)**
- **Smoothing to Remove Noise (J2 Model with Drag)**
- **Outlier Detection/Removal (Test on Angular Momentum)**
- **Bad Data Point Count/Residuals for Quality**
- **Mean Residual Error (Exclusive of Bias) – approximately 10M**

Source of Ephemeris



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Ephemeris Data Smoothing

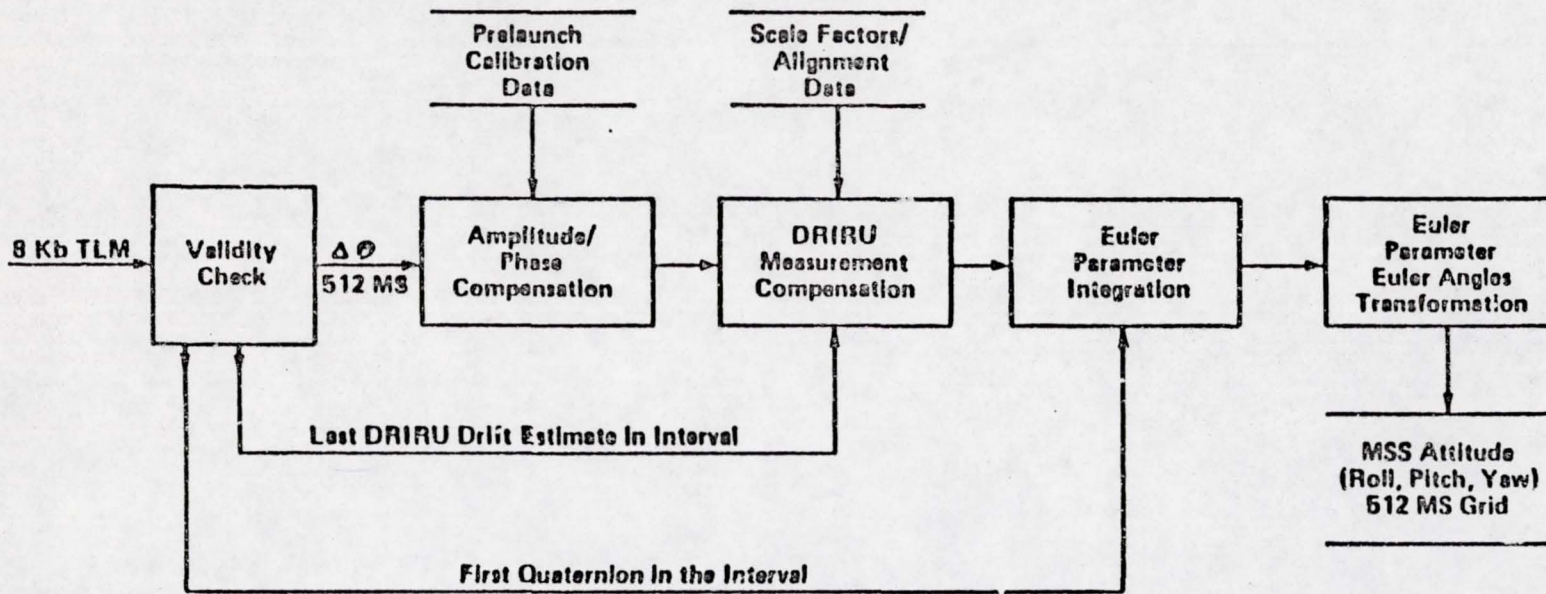


● Outlier rejection criterion: angular momentum error $> 0.002 \times$ angular momentum

Attitude Data Processing

- **Received from the Spacecraft**
 - **Quaternions on a 4.096 Second Grid**
 - **Angular Increments (Pitch, Roll, Yaw) on a 0.512 Second Grid, Filtered (Frequencies less than 0.5 Hz)**
 - **Data in Earth Centered Inertial Coordinate System**
- **Required Processing**
 - **Phase and Amplitude Compensation for Angular Increments**
 - **Limit Checking to Remove Outliers (Transmission Error)**
 - **Integration of Euler Parameters (Rates from Angular Increments)**
 - **Quality Checks**

Data Flow in MSS Attitude Data Processing



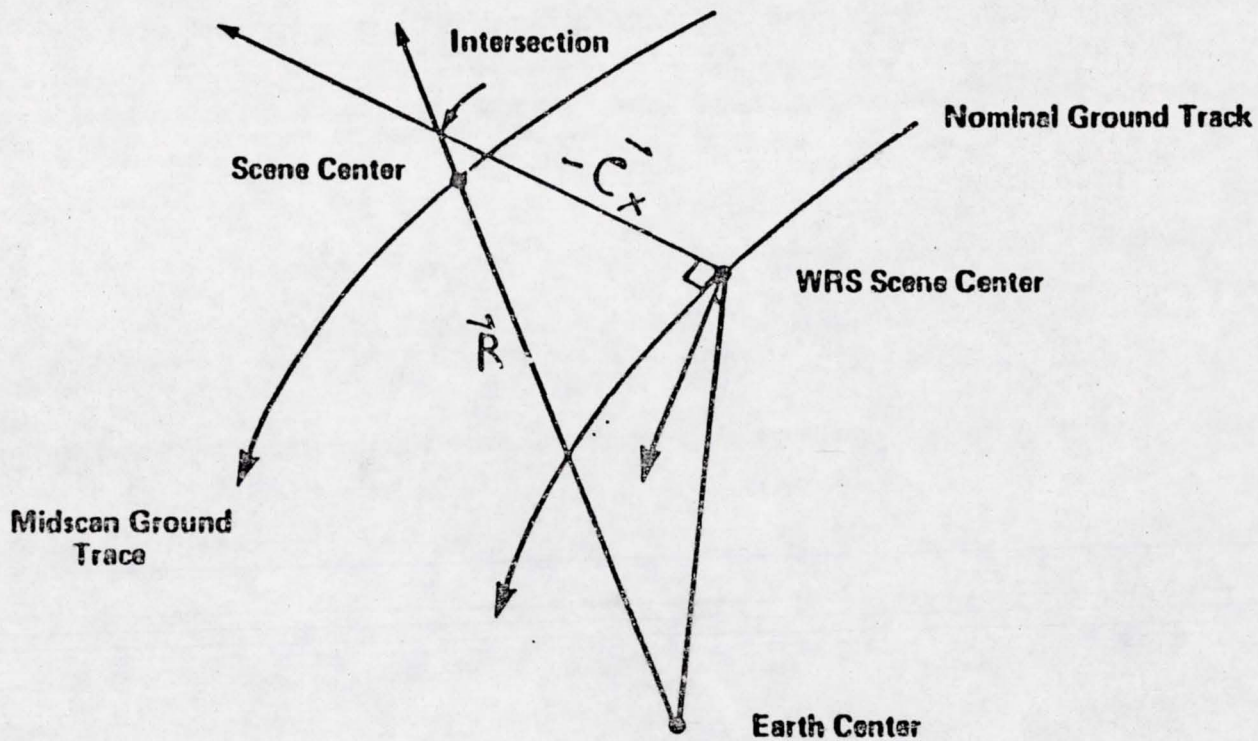
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Example of Quaternion
In Small Angle Approximation

$$\begin{aligned}
 q_1 &\approx \frac{1}{2} \theta_P \\
 q_2 &\approx \frac{1}{2} \theta_R \\
 q_3 &\approx \frac{1}{2} \theta_Y \\
 q_4 &\approx 1.0
 \end{aligned}$$

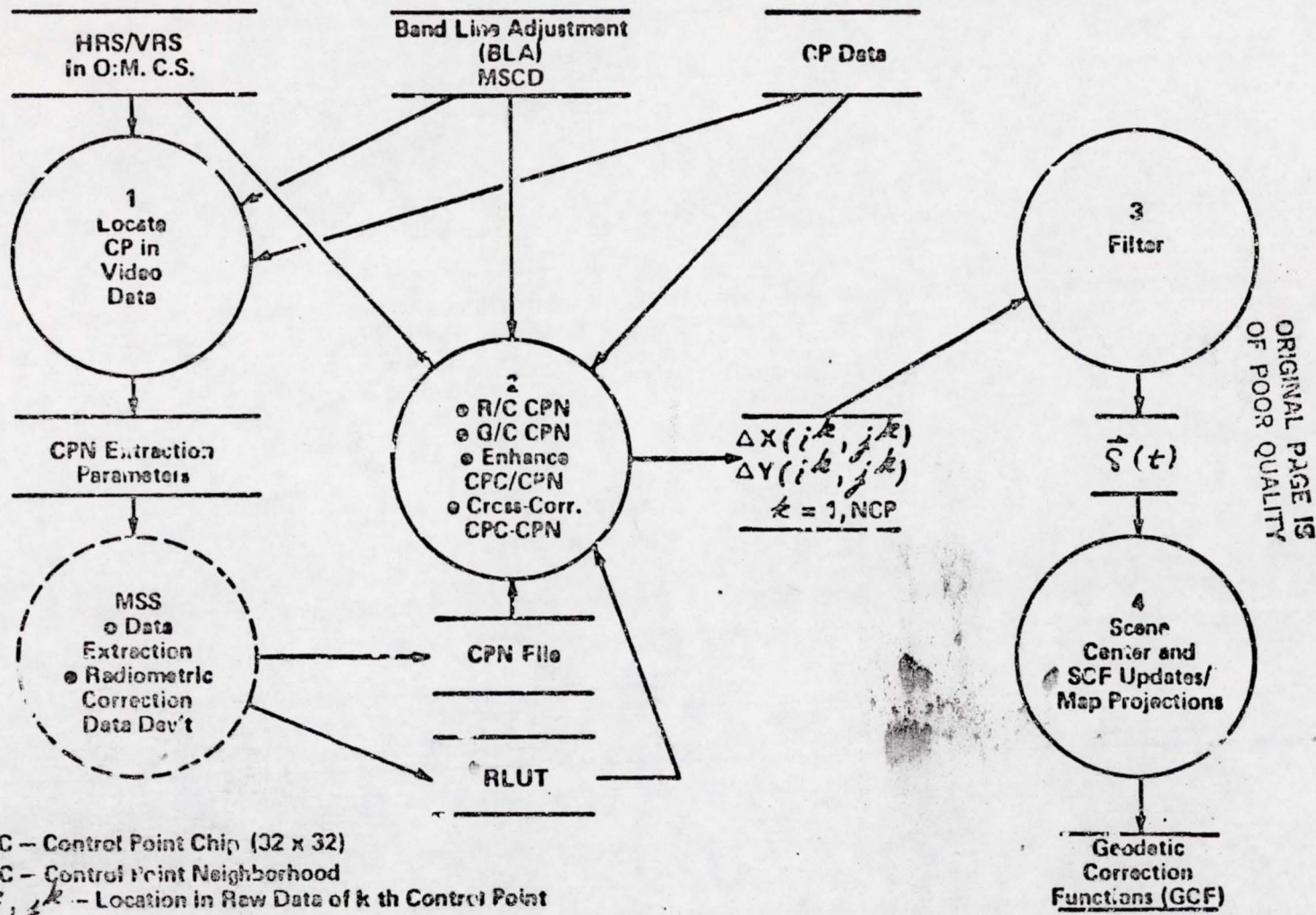
Scene Center Calculation

- Scene Center Time: Time at Which X - Axis of Standard Coordinate System Is Intersected by a Radius Vector Through Midscan Ground Trace



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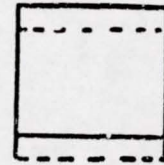
MSS Data Flow In GCDG



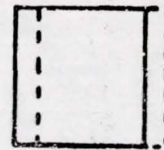
Properties/Effects Of $\vec{\delta}$ For Landsat-D MSS (Single Scene)

• Constant $\vec{\delta}$

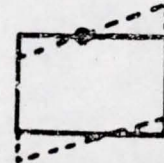
- $\delta\alpha, \delta\theta\rho$ - Along Track Displacement



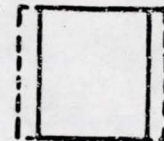
- $\delta\beta, \delta\theta\eta$ - Cross Track Displacement



- $\delta\theta\gamma$ - Along Track Asymmetry



- $\delta\eta$ - Cross Track Asymmetry (Magnification)



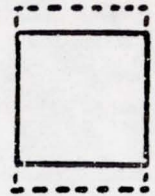
- Asymmetries Almost Linear and Odd Functions of X
- Displacements Large
- Asymmetries Small But Detectable (0.25 - 1.5 Pixel)

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$\vec{\delta}$ Rates - All Asymmetric Image Distortions

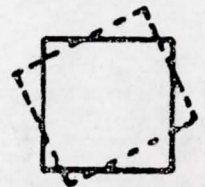
- $\delta \dot{\alpha}, \delta \dot{\theta}_p$

- Along Track Asymmetry
(Magnification)



- $\delta \dot{\beta}, \delta \dot{\theta}_r$

- Cross Track Asymmetry
(Rotation or Yaw Plus Cross Track Skew)



- $\delta \dot{\theta}_y$

- Growth of Along Track Skew



- $\delta \dot{\eta}$

- Growth of Cross Track Magnification



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● Asymmetries Almost Linear and Odd Functions of Y.

● Asymmetries Marginally Detectable (Less Than 0.1 Pixel, 90%)

Filter Variables

- $\delta\alpha$ and $\delta\theta_p$ Hard to Separate — Represented by Single Variable — δSAT , For Along Track Translations
- $\delta\beta$ and $\delta\theta_R$ Hard to Separate — Represented by Single Variable δSCT , for Cross Track Translations
- Hence a 6-Variable Filter:

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Bias Name	Effect in Image	90% Error at Worst Point (Pixels)
δSAT	Along Track Shift	> 10
δSCT	Cross Track Shift	> 10
$\delta\alpha$	Cross Track Magnification	~ 0.1
$\delta\theta_y$	Along Track Skew	~ 0.8
$\delta\dot{SAT}$	Along Track Magnification	~ 0.1
$\delta\dot{SCT}$	Cross Track Skew	~ 0.1

Basis Of The MSS Filter

- Linearity of SCF with $\vec{\delta} = (\delta\alpha, \delta\beta, \delta\gamma, \delta\sigma_p, \delta\theta_x, \delta\theta)$

- Availability of Analytic Partial Derivatives

$$\mu_{x\ell}(i, j) = \partial X(i, j) / \partial \delta_\ell$$

$$\mu_{y\ell}(i, j) = \partial Y(i, j) / \partial \delta_\ell$$

- $\Delta X(i, j) = \sum_{\ell} \mu_{x\ell}(i, j) \cdot \delta_\ell(i, j)$
- $\Delta Y(i, j) = \sum_{\ell} \mu_{y\ell}(i, j) \cdot \delta_\ell(i, j)$

} Time Simply Related to ij

- Above Suggests an Optimal Least Squares Filter, Based on Function:

$$NCP \left(\overline{\epsilon_x^2} / \sigma_x^2 + \overline{\epsilon_y^2} / \sigma_y^2 \right) \equiv F(\vec{\delta}) / \sigma_x^2$$

Where $\overline{\epsilon_g^2} = \frac{1}{NCP} \sum_{k=1}^{NCP} \left\{ \Delta g^k - \sum_{\ell} \mu_{g\ell}(i^k, j^k) \cdot \delta_\ell(i^k, j^k) \right\}^2 ; g=x, y$

and $\sigma_g^2 = \text{Variance of Image Noise/Measurements}$

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CHOOSING THE OPTIMAL VARIABLE SET

- X CORRECTION AND Y CORRECTION ARE ALMOST INDEPENDENT
- X CORRECTING BIASES: δ_{CT} , δ_R , $\delta_{\dot{CT}}$
 - COMPUTE X-QUALITY FOR
 1. ALL 3 VARIABLES
 2. δ_{CT} AND EACH OF 2 INDIVIDUALS
 3. δ_{CT} ALONE
 - RETAIN VARIABLE SET THAT PRODUCES BEST QUALITY
- Y CORRECTING BIASES: δ_{AT} , δ_{e_y} , $\delta_{\dot{AT}}$
 - PROCEED AS FOR X, USING Y-QUALITY
 - RETAIN VARIABLE SET THAT PRODUCES BEST QUALITY
- REPORT VARIABLE SET USED, X- AND Y- QUALITY

Quality Of Correction

- Definition of X-Quality:
Residual Error in X at Worst Points in Scene (Corners)
- Definition of Y-Quality:
Residual Error in Y at Worst Points in Scene (Corners)
- Semi-Analytic Functions for X and Y Quality
Deduced for Given NCP and Spatial Distribution of
CPs — Also Depend on σ_x, σ_y Directly
- X- and Y-Quality Deduced Dynamically and Used to
Select Optimal Variable Set
- Outliers Removed in Usual Way

FILTER OUTPUTS

- STATE VECTOR ($\vec{\delta}$)
- COVARIANCE MATRIX
- ALONG TRACK, CROSS TRACK QUALITY MEASURES
- RESIDUALS AT CONTROL POINTS
- OUTLIER ENUMERATION
- VARIABLES ESTIMATED

GEOMETRIC CORRECTION CALIBRATION

- PURPOSE: IMPROVE SYSTEMATIC CORRECTION FOR SCENES LACKING CONTROL POINTS
- PARAMETERS TO BE CALIBRATED
 - MIRROR MODEL PARAMETERS
 - INSTRUMENT MISALIGNMENTS
 - IMAGE NOISE MATRIX ELEMENTS
 - A PRIORI STATISTICS OF THE SPACECRAFT PARAMETER BIASES
- CALIBRATION DATA - LONG TERM MEANS AND VARIANCES
 - MEAN AND VARIANCE OF LINE LENGTHS
 - MEANS AND VARIANCES OF SPACECRAFT PARAMETER BIASES
 - RESIDUALS AT CONTROL POINTS - MEANS AND VARIANCES OF RESIDUALS SORTED ON CROSS TRACK LOCATION

GEOMETRIC CORRECTION CALIBRATION (CONTINUED)

• CALIBRATION APPROACH

- EFFECTS MIXED ON THE GROUND - SELECT SINGLE PARAMETER TO BE UPDATED
- NON-ZERO MEANS OF SPACECRAFT PARAMETER BIASES POINT TO MISALIGNMENTS, MIRROR AMPLITUDE OR MIRROR VELOCITY ERRORS
- VARIANCES OF SPACECRAFT PARAMETER BIASES PROVIDE STATISTICS FOR SELECTING OPTIMAL VARIABLE SET
- CROSS TRACK PATTERNS IN THE RESIDUALS POINT TO MIRROR VELOCITY PROFILE ERRORS
- DEVIATION OF MEAN LINE LENGTH FROM NOMINAL - MIRROR SPEED CORRECTION - NO IMPACT

Closing Remarks

John Barker

LOA List of Acronyms

AAT	Archival Ancillary (Data) Tape	AT	Acceptance Test
ACCA	Automatic Cloud Cover Assessment	ATL	Applications Technology Laboratory
ACE	Attitude Control Electronics	ATM	Antenna Test Model
ACS	Attitude Control System	ATM	Apollo Telescope Mount
ACT	Application Concept Test	ATP	Acceptance Test Plan
A/D	Analog to Digital	ATS	Applications Technology Satellite
AODP	See ANDP	AWG	American Wire Gauge
AODS	Applications Developmental Data System		
ADFS	Automated Digital Facsimile System	BARDJA	Boom Antenna Retention Deployment and Jettison Assembly
ADL	Applications Development Laboratory	BAT	Bench Acceptance Test
ADP	Automatic Data Processing	BB	Build Baseline
ADPE	Automatic Data Processing Equipment	BCU	Bus Coupling Unit
A&DS	Aerospace and Data Systems	BDF	Block Data Format
ADS	Angular Displacement Sensor or Angle Detector Sensor	BER	Bit Error Rate
ADT	Ancillary Data Tape	BESS	Biological Experiment Scientific Satellite
AEM	Applications Explorer Mission	BFR	Browse Film Recorder
AFGWC	Air Force Global Weather Central	BIC	Band Interleaved by Cylinder
AFOS	Automation of Field Operations and Services	BIL	Band Interleaved by Line
AFPRO	Air Force Plant Representative Office	BIP	Band Interleaved by Pixel
AG	Archive Generation	BIW	Band Interleaved by Word
AGC	Automatic Gain Control	BOL	Beginning of Life
AGE	Aerospace Ground Equipment	BOS	Beginning of Scan
AGS&PO	Aerospace Group Strategic Planning and Programs Office	BOT	Beginning of Tape
Ahr	Ampere - hour	B&P	Bid and Proposal
ALU	Algorithm Logic Unit	BPA	Bus Protection Assembly
AMS	Attitude Measurement System	bpi	Bits per Inch
AN	Applications Notice	BPI	Bytes per Inch
ANCP	See ANDP	bps	Bits per Second
ANDP	Ancillary Data Calculation Process	BPS	Bytes per Second
ANSI	American National Standards Institute	BSE	Broadcast Satellite Experimental
ANT	Ascending Node Table	BSQ	Band Sequential
AO	Announcement of Opportunity	BSR	Back Surface Radiator
AOIPS	Atmospheric and Oceanographic Image Processing System	BTC	Bench Test Cooler
AOP	Advanced Onboard Processor	BTCE	Bench Test and Calibration Equipment
AOS	Acquisition of Signal	BTE	Bench Test Equipment
AP	Applications Processor	B/U	Backup
AP	Array Processor	B&W	Black and White
APFO	Aerial Photography Field Office		
APL	Applied Physics Laboratory (John Hopkins Univ.)	CAL	Configured Articles List
APS	Antenna Positioning System	CAL	Calibration
ASCII	American Standard Code for Information Interchange	CARETS	Central Atlantic Regional Ecological Test Site
ASPR	Aerospace Strategic Program Representation	CASH	Catalog of Available and Standard Hardware
ASPR	Armed Services Procurement Regulations	CAT	Catalog
ASR	Automatic Send/Receive	CCA	Cloud Cover Assessment
AST	Asynchronous System Trap	CCB	Configuration Control Board
ASVT	Applications System Verification and Transfer Project	CCC	Camera Controller Combiner

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CCD Charge Coupled Device
 CCL Closed Circuit Loop
 CCM Color Composite Master
 CCN Contract Change Notice
 CCP Cloud Cover Assessment Process
 CCT Computer Compatible Tape
 CCT-A CCT Containing Partially-Corrected Data
 CCT-AH CCT Containing Partially-Corrected MSS Sensor Data
 CCT-AT CCT Containing Partially-Corrected TM Sensor Data
 CCT-B CCT Produced by ADOS
 CCT-P CCT Containing Fully-Corrected Data
 CCT-PM CCT Containing Fully-Corrected MSS Sensor Data
 CCT-PT CCT Containing Fully-Corrected TM Sensor Data
 CDD Cartridge Removable Diablo Disk Drive
 C&DH Command and Data Handling
 CDHS Command and Data Handling System
 CDHSS Command and Data Handling System Simulator
 CDHSS I/U CDHSS Interface Unit
 CDR Conceptual Design Review
 CDR Critical Design Review
 CDRB Conceptual Design Review Board
 CDRL Contract Data Requirements List
 CEM Controlled Environment Module
 CFOV Clear Field-of-View
 CG Center of Gravity
 CI Configuration Item
 CLD Cloud
 CLL Corrected Line Length
 CM Center of Mass
 C.M. Configuration Management
 CMD Command
 CMI Configuration Management Instruction
 CMM Command Memory Management
 CMO Configuration Management Office
 COBOL Common Business Oriented Language
 COMP Computer
 C.P. Center of Pressure
 CP Communication Processor
 CP Control Point
 CPC Control Point Chip
 CPCI Computer Program Configuration Item
 CPD Control Point Directory
 CPDS Computer Program Design Specification
 CPD-U Control Point Directory (Candidate for Permanent File)
 CPG Correction and Product Generation Software
 CPL Control Point Library
 CPL-U Control Point Library (Candidate for Permanent File)
 cpm Cards per Minute
 CPM Computer Personality Module
 CPN Control Point Neighborhood
 CPN-G Control Point Neighborhood for Geodetic Corrections
 CPN-L Control Point Neighborhood for Library Maintenance
 CPN-M Control Point Neighborhood for MSS
 CPN-T Control Point Neighborhood for TH
 CPPT CZCS Preprocessor Performance Tape
 CPR Cloud Physics Radiometer
 CPU Central Processing Unit

CR Card Reader
 CRC Cyclic Redundancy Check
 CRIS Cosmic Ray Ionization Spectrometer
 CRT Cathode Ray Tube
 CSA Cropping, Subsampling and Averaging
 CSC Computer Sciences Corporation
 CSE Contractor Supplied Equipment
 CSF Control and Simulation Facility
 CSS Coarse Sun Sensor
 CU Central Unit
 CY Calendar Year
 CZCS Coastal Zone Color Scanner

 D/A Digital-to-Analog
 DAS Data Base Administration Subsystem
 DAS³ De-Centralized Automated Service Support System
 DB Data Base
 DBIP Data Base Interface Process
 dBi Antenna gain in decibels referenced to an Isotropic Antenna

 dBm Power in decibels referenced to one millimeter
 DBMS Data Base Management System
 DBMS-10 DEC-10 System Software for Data Base Management
 DC Direct Current
 DCP Data Collection Platform
 DCS Data Collection System
 DCST Data Collection System Tape
 DDG Digital Display Generator
 DDI Digital Data Interconnect
 DDL Data Description Language
 DDP Digital Data Processor
 DDP-C Controlled Environment Module DDP
 DDP-W Wire-Wrapped DDP
 DDR Detailed Design Review
 DDRB Detailed Design Review Baseline
 DEC Digital Equipment Corporation
 DEC-10 DEC-10 Computer
 DEC-20 DEC-20 Computer
 DECnet Digital Equipment Corporation Communications Network
 DECOM Decommutator
 DECOM Decommutation Hardware Device
 DEMJX Demultiplexer
 DFP Data Formatter Processor
 DFS/ADFS Digital Facsimile System/Automated Digital Facsimile System

 DIAL Digital Image Analysis Laboratory
 DICOMED Film Recorder Vendor
 DID Digital Image Data
 DIP Dual Inline Package
 DIPS Digital Image Processing System
 DKIO Large Image Access Routines
 D/L Downlink
 DMA Direct Memory Access
 DMF Data Management Facility
 DML Data Management Language
 DML Data Manipulation Language
 DMS Data Management System

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DMSP	Defense Meteorological Satellite Program	EIA	Electronic Industries Association
DOC	Data Operations Control	ELE	Elevation at Entry
DOO	Department of Defense	ELS	End-of-Line Sync
DOO	Depth of Discharge	ELX	Elevation at Exit
DOI	Department of the Interior	EMC	Electromagnetic Compatibility
GOI/EDC	Department of the Interior/EROS Data Center	EMI	Electromagnetic Interference
DOMSAT	Domestic Communications Satellite	ENA/DISA	Enable/Disable
DPM	Drafting Practices Manual	EOB	End-of-Buffer
DPR	Design Problem Report	EOF	End-of-File
DPS	Data Processing System	EOL	End-of-Life
DPS	DRRTS Process Software	EOM	End-of-Mission
DPSE	DRRTS Process Software Executive	EOP	Earth Observatory Program
DPU	Digital Processing Unit	EGP	End-of-Process
DR11C	Programmed Input Output Interface Device for DEC Unibus	EORT	End-of-Roll Target
DR70	Direct Memory Access Interface Device for DEC Massbus	EOS	End-of-Scan
DR780	Direct Memory Access Interface Device for DEC VAX-11/780	EOS	Earth Observation Systems
DRIRU	Dry Rotor Inertial Reference Unit	EOS	Earth Observations Satellite
DRRTS	Data Receive, Record and Transmit System	EOS	End-of-Set
DS	Dimension (Telephone) System	EO&SP	Earth Observatory and Shuttle Programs
DSC	Data Collection System	EOT	End-of-Tape
DSCS	Defense Satellite Communications System	EOV	End-of-Volume
DSCS	Desk Side Computer System	EPA	Environmental Protection Agency
DSI	Digital Subsystem Interface Unit	EPC	Electrical Power Conditioner
DSL	Data Service Laboratory	EPI	Euler Parameter Integration
DSM	Downlink Synchronization Module	EPS	Electrostatic Plotting Software
DSSCI	Data Stripper-Serial Controller Interface	ER	Equipment Room
DSU	Digital Switching Unit	EREP	Earth Resources Equipment Package
DTD	Digital Terrain Data	EROS	Earth Resources Observation System or Satellite
DTM	Digital Terrain Map	ERS	Earth Resources Survey
DTG	Digital Tape Generation	ERTS	Earth Resources Technology Satellite
DTR	Daily Test Report	ESA	European Space Agency
DTS	Digital Transmission System	ESR	Equipment Service Report
DV	Digital Voltmeter	ESTEC	European Space Research and Technology Center
DX20	DEC Peripheral Interface Device	EU	Expander Unit
DXFP	Data Extraction and Formatting Process	EVA	Extra-Vehicular Activity
EAGE	Electrical Aerospace Ground Equipment	EVAL	Earth Viewing Applications Laboratory
EBCDIC	Extended Binary Coded Decimal Interchange Code	EWO	Engineering Work Order
EBR	Electron Beam Recorder	FAIRS	Full Aperture Infrared Source
EBRIC	Electronic Beam Recorder Image Correction	FAAO	Financial and Administrative Operations
ECC	Error Correction Capability (HCCR)	FAS	Foreign Agricultural Service
ECEF	Earth-Centered-Earth-Fixed	FCS	File Control Service
ECI	Earth-Centered-Inertial	FDD	Fixed (Cartridge) Diablo Disk (Drive)
ECL	Emitter Coupled Logic	FDR	Final Design Review
EDC	EROS Data Center	FFP	Federation of Functional Processors
EDIPS	Electronic Digital Processing System	FGS	Fine Guidance System
EDIPS	EDC Digital Image Processing System	FHST	Fixed-Head Star Tracker
EDP	Electronic Data (Digital) Processing	FID	Final Instrument Definition
EDPS	Electronic Data Processing System	FIFO	First-In, First-Out
EED	Electro-Explosive Device	FIPS	Federal Information Processing Standards
EF	Earth Fixed Coordinate System	FM	Frequency Modulation
EGRET	Explorer Gamma Ray Experiment Telescope	FM	Flight Model
EGSE	Electrical Government Supplied Equipment	FMEA	Failure Mode and Effects Analysis
EI	Engineering Instruction	FMS	Flight (Segment) Management Subsystem
		FO	Flight Operations
		FOC	Faint Object Camera
		FORTAN	Formula Translation

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FOS Field Operations Service
 FOS Flight Operations Subsystem
 FOS Faint Object Spectrograph
 FOV Field-of-View
 FPA Focal Plane Assembly
 FPG Final Product Generation
 FPP Floating Point Processor
 FPS Focal Plane Structure
 FRD Facilities Requirement Document
 FRS Film Recorder System
 FRUSA/HASP Flexible Roll-Up Solar Array/Hardened Solar Power System
 FS Flight Segment
 ESCM Federal Supply Code For Manufacturers
 FSDF Flight Segment Development Facility
 FSEC Fairchild Space and Electronics Company
 FSK Frequency Shift Keying
 FSS Flight Scheduling Subsystem
 FSS Flight Segment Simulator
 FSS Flight Support System
 FSS Fine Sun Sensor
 FSS S/W Flight Segment Simulator Software
 FT Fourier Transform
 FTS Federal Telephone System
 FW Fiscal Week
 FY Fiscal Year
 FYI For Your Information

 G/C Geometric Correction
 GCO Geodetic Correction Data or Geometric Correction Data
 GCOG Geodetic Correction Data Generation
 GCM Geometric Correction Matrices
 GCM Geometric Correction Matrix
 GCO Geometric Correction Operator
 GCOVS GCO Verification System
 GCP Geodetic Control Point
 GCP Ground Control Point
 GDHS Ground Data Handling System
 GDT Graphics Display Terminal
 GE General Electric
 GE70 GE Interface Device for DR780
 GECP Geometric Correction Process
 GEOREF Geographic Reference
 GES Ground Electronic Specification
 GETSCO General Electric Technical Service Company
 GFE Government Furnished Equipment
 GFIT Goddard Film Inventory Tape
 GFP Government Furnished Property
 GHIT Goddard HDT Inventory Tape
 GHz Gigahertz (10^9)
 GI General Instruction
 GIA Government Inspection Agency
 GM General Manager
 GMF GCO Microcode File
 GMP Geometric Correction Matrix Calculation Process
 GMS Ground (Segment) Management Subsystem

GMT Greenwich Mean Time
 GOES Geostationary Operational Environmental Satellite
 GPC General Purpose Console
 GPE Ground Processing Equipment
 GPIB General Purpose Information Processor
 GPS Global Positioning System
 GPT General Purpose Transformation
 CRE Gamma Ray Explorer
 GRFP Graphite Filled Epoxy
 GS Ground Segment
 GSE Ground Support Equipment
 GSFC Goddard Space Flight Center
 GSSS Ground Support System Software
 GSTDN Ground Spaceflight Tracking and Data Network

 HAAT Header, Ancillary, Annotation, Trailer
 HATT-L HATT for Library Maintenance
 HAC HDDR Assignment and Control
 HAT Header Annotation, Trailer
 HAL High-Order Aerospace Language
 HCMM Heat Capacity Mapping Mission
 HD HDT Duplication
 HDDR High Density Digital Recorder
 HDDT High Density Digital Tape
 HDE HDT-R Directory Extractor
 HDT High Density Tape
 HDT-A HDT-Archive Format (Partially corrected radiometrically but not geometrically)
 HDT-AM HDT-A for MSS Sensor Data
 HDT-AMC Copy of HDT-A for MSS Sensor Data
 HDT-ATC HDT-A for TM Sensor Data
 HDT-ATC Copy of HDT-A for TM Sensor Data
 HDT-I HDT (Data) Interval
 HDT-P HDT-Product Format (Fully corrected)
 HDT-PT HDT-P for TM Sensor Data
 HDT-PTC Copy of HDT-P for TM Sensor Data
 HDT-R High Density Tape Recorder
 HDT-R Raw Data as recorded in DRRTS
 HDT-RM HDT-R for MSS Sensor Data
 HDT-RT HDT-R for TM Sensor Data
 HDT-S HDT Recorded at White Sands
 HDT-SM HDT-S for MSS Sensor Data
 HDT-ST HDT-S for TM Sensor Data
 HgCdTe Mercury Cadmium Telluride
 HIPO Hierarchy Input Process Output
 HPRR High Resolution Film Recorder
 HSC High Speed Control Element
 HSI High Speed Interface
 HV Host Vehicle (Landsat-D)
 H/W Hardware
 Hz Hertz (cycles per second)

 IAC Image Analyzer Console
 IAP Integrated Analysis Plan
 IAT Image Analysis Terminal
 IAT Image Annotation Tape
 IB Integration Baseline

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ICCD Intensified Charge Coupled Device
 ICD Interface Control Document
 ICS Image Correction Support Software
 ICS Interactive Computer Simulator
 ID Identification
 IDA Image Data Acquisition
 IDB Identification Burst
 IDBS International Data Base Systems
 IDS Image Data System
 IDT Investigation Definition Team
 IDT Image Display Terminal
 IDT Industrial Data Terminal Corporation
 IDT Image Data Transmission
 I/F Interface
 IF Intermediate Frequency
 IFD In-Flight Disconnect
 IFJV Instantaneous Field-of-View
 IG Initial Gap
 IGF Image Generation Facility
 IIGS Initial Image Generation Subsystem
 IIRV Improved Inter-Range Vectors
 IIS (I²S) International Imaging Systems
 IM Information Management
 IM Instrument Module
 IMPAC Image Processing and Analysis Center
 IMS Information Management Subsystem
 IMSC Information Management Subsystem Computer
 IMSFCC Information Management Subsystem FFP Control Computer
 IMU Image Memory Unit
 InSb Indium Antimonide
 INTRALAB Information Transfer Laboratory
 I/O Input/Output
 IPC Initial Product Creation
 IPCS Information Production Control System
 IPD Information Processing Division
 IPF Image Processing Facility
 ips Inches per Second
 IPS Image Processing Subsystem
 IPS-1 IPS String No. 1 Computers
 IPS-2 IPS String No. 2 Computers
 IPSC IPS Computer
 IQL Interactive Query Language
 IR Infrared
 IRB Integrated Requirements Board
 IR&D Independent Research and Development
 IRD Interface Requirements Document
 IRFPA Infrared Focal Plane Assembly
 IRG Inter-Record Gap
 IRIG Inter-Range Instrumentation Group Time Code
 IRIG-A IRIG Time Code Series A
 IRP Infrared Photometer
 IRO Interrupt Request
 IRU Inertial Reference Unit
 IS Input Subsystem
 ISA Instrument Standard of America
 ISAM Index Sequential Access Method

ISM Interface Switching Module
 ISS Image Generation Facility Software Subset
 ISU Input Scanner Unit
 IT Integration Test
 I&T Integration and Test
 ITD Inception-to-Date
 ITD Incurred-to-Date
 ITP Integration Test Plan
 IU Interface Unit
 IUE International Ultraviolet Explorer
 IUS Interim Upper Stage
 JPL Jet Propulsion Laboratory
 JSC Johnson Space Center
 K A Thousand
 K 1024 (Memory Usage Only)
 Kb Kilobit
 KB Kilobyte
 Kbps Kilobits per Second
 KBPS Kilobytes per Second
 KCRT Keyboard Cathode Ray Tube
 KL10 CPU for DEC-10 Computer
 km Kilometer
 KS Key Station
 KSA Ku-band Single Access
 KSC Kennedy Space Center
 KW Kilowords
 LA36 DEC Hardcopy Terminal
 LACIE Large Area Crop Inventory Equipment
 LANDSAT Land Satellite
 LaRC Langley Research Center
 LAS Landsat-D Assessment System
 LAT Latitude
 LBP Library Build Process
 LBR Laser Beam Recorder
 LCP Left-Hand Circularly Polarized
 LED Light-Emitting Diode
 LFC Left-Fill Count
 LIQU Large Image Display Utility
 LIFO Last-In, First-Out
 LLA Adjusted Line Length
 LLC Line Length Code
 LM Library Maintenance
 LM Line Monitor
 LMM Landsat Mission Management
 LMSC Lockheed Missile and Space Corporation
 LOE Level of Effort
 LONG Longitude
 LOS Line of Sight
 LOS Loss of Signal
 LPC Longitudinal Parity Check
 LPM Line Point Marker
 LPM Lines Per Minute
 LPM Load Point Marker
 LRA Laser Retroreflector Array

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LRC Longitudinal Redundancy Check
 LRD Laser Retroreflector
 LSB Least Significant Bit
 LSD Landsat-0
 LS3 Landsat 3
 LTC Light Transfer Characteristics
 LTTS Long-Term Tape Storage Facility
 LTU Line Test Unit
 LUN Logical Unit Number
 LV Launch Vehicle

 M Mega-
 M Million
 MA Multiple Access
 MACS Modular Attitude Control System
 MAG MSS Archival Product Generation
 MAP Macro Array Processor
 MASSBUS High Speed Bus for DEC Equipment
 MATSCO Management and Technical Services Company
 Mb Megabit
 MB Megabyte
 MBA MASSBUS Adaptor
 Mbps Megabits per Second
 MCC Mission Control Center
 MCCA Manual Cloud Cover Assessment Package
 MCR Monitor Console Routine
 MCTF Mission Contractor Test Facility
 M&DO Mission and Data Operations
 M&DOD Mission and Data Operations Directorate
 MDM Multiplex-Demultiplex
 MDP Master Data Processor
 MEM Module Exchange Mechanism
 MERITS Marshall Earth Resources Information Transfer System
 METSAT Meteorological Satellite
 MFB Major Frame Buffer
 MFD Master File Directory
 MFS Major Frame Synchronization
 MGSE Mechanical Government Supplied Equipment
 MHS MSS/HDDR Service
 MHW Multi-Hundred Watt
 MHz Megahertz (10⁶)
 MIF Master Information File
 MIP Management Information Process
 MIPS MSS Image Processing System
 MIPS Mega-Instructions per Second
 MIS Mission Interface Subsystem
 MIT Master Information Table
 MJF Major Frame
 mm Millimeter
 M Minutes
 MWF Mission Management Facility
 MWFCC Mission Management Facility Control Computer
 MWF-M MSS Mission Management Facility
 MWF-T TM Mission Management Facility
 MMS Mission Management System
 MMS Multi-Mission Modular Spacecraft

MMU Memory Management Unit
 MFS Minor Frame Synchronization
 MSO Maintenance and Operations
 MODEM Modulator/Demodulator
 MOI Moments of Inertia
 MOL Manned Orbiting Laboratory
 MOM Mission Operations Manager
 MOPA Mega-Operations per Second
 MOR Mission Operations Review
 MOU Memorandum of Understanding
 MPP MSS Preprocessor
 MPS Mission Planning System
 MPS Modular Power Subsystem
 MPT Maximum Power Tracker
 MPY Multiply
 MRA Maintenance Requirements Analysis
 MRAM Maintenance Requirements Analysis Matrix
 MRC Master Reference Cube
 MRS Module Reference System
 MS Mirror Sweep
 MSB Most Significant Bit
 MSC Manned Space Center
 MSCD-M MSS Mirror Scan Correction Data
 MSCD-T TM Mirror Scan Correction Data
 MSCO Mission Support Coordination Office
 MSC Matrix Switch Control
 MSEC Millisecond
 MSFC Marshall Space Flight Center
 MSR Monthly Status Review
 MSS Module Support Structure
 MSS Multispectral Scanner
 MSS-A MSS Archival Data
 MSW Matrix Switch
 MT Magnetic Tape
 MTBF Mean Time Between Failures
 MTF Modulation Transfer Function
 MTL Material
 MTM Mechanical Test Module
 MTP MSS Telemetry Processor
 MTR Mean Time to Repair
 MTU Magnetic Tape Unit
 MJX Multiplexer
 MW Megawatts

 N₂ Purified and Filtered Gaseous Nitrogen
 N/A Not Applicable
 NAK Negative Acknowledgement
 NAPPS Nimbus/AEM Preprocessor System
 NASA National Aeronautics and Space Administration
 NASCOM NASA Communications Network
 NASTRAN NASA Structural Analysis (Program)
 NASTRAN NASA Transient Analysis System
 NBR Narrow Band Tape Recorder
 NCC National Climatic Center
 NCC Network Control Center
 NCCS Network Control Center Subsystem
 NCIC National Cartographic Information Center

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ND Networks Directorate
 NDF Neutral Density Filter
 NDPF NASA Data Processing Facility
 NDS Navigation Data Satellite
 NDS Navigational Development Satellite
 NESS National Environmental Satellite Service
 NMI NASA Management Instructions
 NOAA National Oceanic and Atmospheric Administration
 NOCC Network Operations Control Center
 NOSS National Oceanographic Satellite System
 NRC Nuclear Regulatory Commission
 NRZ Non-Return to Zero
 NRZI Non-Return to Zero Incrementing
 NRZ-L Non-Return to Zero-Level
 NSCI NASA Serial Controller for Input (now SPDI)
 NSCO NASA Serial Controller for Output (now PSDO)
 NSSC-1 NASA Standard Spacecraft Computer - Model 1
 NSSDC National Space Science Data Center
 NTR New Technology Representative
 NTSC National Television System Committee
 NTTF Network Test and Training Facility
 NTT NASA Tracking and Telemetry Facility

OAO Orbital Astronomy Observatory
 OAO OAO Corporation
 OAS Orbit Adjust Subsystem
 OBC Onboard Computer
 OBP Onboard Processor
 OCB Operational Configuration Baseline
 OCC Operations Control Center
 OCD Operator Control and Display
 OCG Orbit Computations Group
 OCR Optical Character Recognition
 ODF Orbit Determination Facility
 ODP Online Display Process
 ODT Online Debugging Tool
 O&M Operations and Maintenance
 OFLS Offline System
 ONLS Online System
 OPS Operations
 O/S Operations Supervisor
 OS Operating System
 OSCD Orbital Support Computing Division
 OSO Orbiting Solar Observatory
 OSR Optical Solar Reflector
 OSS Office of Space Science
 OSS Operating System Software
 OTA Optical Telescope Assembly
 OTDA Office of Tracking and Data Acquisition

PA Public Address
 PAO Public Affairs Office
 PAM Pulse Amplitude Modulation
 P/ATH Orbital Path
 P/B Playback
 PBX Private Branch Exchang.
 PC Production Control

PC Program Counter
 PC Printed Circuit
 PCB Printed Circuit Board
 PCD Payload Correction Data
 PCD Photon Counting Detector
 PCD-M MSS Payload Correction Data
 PCD-T TM Payload Correction Data
 PCE Pipeline Control Executive
 PCM Pulse Code Modulated
 PCP Product Control Procure
 PCP Program Control Procedure
 PCS Payload Correction Subsystem
 PCU Power Control Unit
 PD Payload Disconnect
 PD Programmable Decommutator
 PDF Programmable Data Formatter
 PDL Program Design Language
 PDP Programmable Digital Processor
 PDP Peripheral Data Product
 PDR Preliminary Design Review
 PDR Problem/Defect Report
 PDSS Precision Digital Sun Sensor
 PDU Power Distribution Unit
 PE Performance Evaluation
 PE Phase Encoded
 P&E Plant and Equipment
 PES Performance Evaluation Subsystem
 PET Predicted Ephemeris Tape
 P/F Protoflight
 PFD Pre-Flight Disconnect
 PFD Protoflight and Flight
 PFI Program Funding Instructions
 PGCOP Product Generation CCT Output Process
 PGHIP Product Generation HDT Input Process
 PGHSM Product Generation HDT-P Simulator
 PGLOP Product Generation LBR Output Process
 PGLSM Product Generation LBR Simulator
 PGM Program Manager
 PGMON Product Generation Pipeline Monitor Process
 PGP Product Generation Process
 PGS Product Generation Subsystem
 P/I Policy/Instruction
 PI Principal Investigator
 PIF Pseudo Image File
 PIGP Pseudo Image Generation Program
 PIL Pixel Interleaved by Line
 PIO Programmed Input Output
 PIP Peripheral Interchange Program
 PIR Program Information Request/Release
 PIXEL Picture Element
 PKG Package Design Specification
 P/L Payload
 PLMCE Post Landsat-D Advanced Concepts Evaluation
 PM Preventive Maintenance
 PM Propulsion Module
 PMS Program Management Budget
 PMD Post-Mortem Dump

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PM/FL	Performance Monitor/Fault Location	RCFP	Radiometric Correction Function Calculation Process
PMM	Program Maintenance Manual	RCHP	Right-Hand Circularly Polarized
PMP	Premodulation Processor	RCP	Registration or Relative Control Point
PMT	Photomultiplier Tube	RCP	Right-Hand Circularly Polarized
PN	Pseudo Noise	RCV	Receive
PO	Project Office	R&D	Research and Development
PO	Purchase Order	RDCP	Radiometric Corrected Process
POCC	Payload Operations Control Center	RDCP	Radiometric Function Calculation Process
POD	Project Operations Directors	RDR	Raw Data Tape
POP	Project Operating Plan	REC	Record
PORTS	Preliminary Operations Requirements and Testing Support	REM	Rocket Engine Module
POWO	Purchase Order Work Order	RF	Radio Frequency
PPL	Photo Processing Lab	RFC	Right-Fill Count
PPL	Preferred Parts List	RFH	Request for Hire
PPO	Program Participation/Opportunities System	RFOV	Resolution Field-of-View
PPS	Photographic Processing Subsystem	RFP	Request for Proposal
PRMIS	Printing Resource Management Information	RH780	Massbus Adaptor for DEC VAX-11/780
PRN	Pseudo Random Noise	RID	Review Item Discrepancy
PRO	Payload Receiving Operations	RIU	Remote Interface Unit
PROM	Programmable Read-Only Memory	RLUT	Radiometric Lookup Table
FRP	Performance Recognition Program	RMS	Remote Manipulator System
PRU	Power Regulator Unit	RMS	Root Mean Square
PS	Polar Stereographic	RMS	Record Management Services
PSDO	Parallel-to-Serial Data Output Device	ROM	Read-Only Memory
PSF	Photo/Shipping Support Facility	ROW	Geographic Frame Reference
PSK	Phase Shift Keying	RPO6	DEC 176 MB Disk or Removable Disk Storage Unit
PSM	Programmable Sync Module	RPO7	DEC 283 MB Disk
PSR	Project Status Review	R/PA	Receiver/Processor Assembly (GPS)
PSU	Power Supply Unit	R&PA	Reliability and Product Assurance
PSU	Power Switching Unit	RPM	Revolutions Per Minute
PVS	Pressure Vessel Spacecraft	RPP	RBV Preprocessor
PWB	Printed Wiring Board	R&QA	Reliability and Quality Assurance
PWN	Pulse Width Modulated	RSE	Receiving Site Equipment
Q&A	Qualification and Acceptance	RSE	Remote Site Equipment
QA	Quality Assurance/Assessment	RSX-11M	Multi-Tasking Operating System Software
QAF	Quality Assessment Film	R/T	Real-Time
QAP	Quality Assessment Process	RTG	Radiosotope Thermoelectric Generator
QAP	Quality Assurance Procedure	RTTS	Real-Time Test System
QAP	Qualification and Acceptance Program	RX	Receive
QC	Quality Control	SA	Single Access
QFP	Quality Assurance Film Generation Process	SA	Solar Array
QIO	Queued Request for Input/Output	SAD	Solar Array Drive
QIO	Queue Input/Output Process	SADAPTA	Solar Array Drive and Power Transfer Assembly
QLD	Quick Look Display	SAIL	Space Applications and Information Library
QLM	Quick-Look Monitor Unit	SARJA	Solar Array Retention, Deployment and Jetison Assembly
QLP	Quick-Look Processor	SB	Stage Baseline
QLPS	Quick-Look Processing System	SBC	Single Board Computer
QPSK	Quadrature Phase Shift Keyed	SBI	Synchronous Backplane Interconnect
QRWO	Quick-Reaction Work Order	SBRC	Santa Barbara Research Center
QSL	Quarter Scan Line	SBS	Space Background Simulator
RAA	Reformatting Ancillary Annotation	SBU	Strategic Business Unit
RAM	Random Access Memory	S/C	Spacecraft
RBV	Return Beam Vidicon	SC	Signal Conditioning
RC	Radiometric Correction	SCA	Signal Conditional Assembly
		SCAMA	Switching, Conferencing and Monitoring Arrangement

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SCCB Software Change Control Board
 SCD Systematic Correction Data
 SCHS Spacecraft Hardware Simulator (MSS Simulator)
 SCI Serial Control Interface
 SCII Serial Control Interface for Input (now SPDI)
 SCIO Serial Control Interface for Output (now PSDO)
 SCL Subcontract Labor
 SCM Systematic Correction Matrix
 SCP Sun Calibration Process
 SCR Scaler Control Register
 SCR Software Change Request
 SC&SU Signal Conditioning and Switching Unit (SU)
 SCT System Control Terminal
 SD Space Division
 SDF Software Development Facility
 SDHS Satellite Data Handling System
 SDISS Satellite Data Ingest and Storage Subsystem
 SDSB Satellite Data Services Branch
 SEAM Software Engineering and Management Program
 Sec Seconds of Arc
 SECO Secondary Electron Conduction Orthicon
 SEOPS Standard Earth Observation Package Satellite
 SEOS Synchronous Earth Observation Satellite
 SI Science Instruments
 SIAT Special Image Annotation Tape
 SICM Science Instrument Central Module
 SIDU Small Image Display Utility
 SIF Simulation Image File
 SIM Simulator
 SIP System Image Preservation
 SIRD Support Instrumentation Requirement Document
 SIU Sectorizer Ingest Unit
 SLAT Spacecraft Location and Attitude Tape
 SLC Scan Line Corrector
 SLER Synch Loss Error Rate
 SLP Source Language Input Program
 SLS Scan Line Sync
 SLS Start-of-Line Sync
 SMA S-Band Multiple Access
 SMA Scan Mirror Assembly
 SMM Solar Maximum Mission
 SM&O Support Maintenance and Operations
 SMP Scan Monitor Pulse
 SMK Software Modification Record
 SMSA Standard Metropolitan Statistical Area
 S/N Signal-to-Noise Ratio
 SNR Signal-to-Noise Ratio
 SOM Space Oblique Mercator
 SOP Standard Operating Procedure
 SOW Statement of Work
 SP Stack Pointer
 SPC Small Peripheral Controller
 SPD DEC Software Product Description
 SPDI Serial-to-Parallel Data Input Device
 SPM Sub-Project Manager
 SPP Special Purpose Processor
 SPR Software Problem Report

SPRD Site Preparation Requirements Document
 SPS Segment Processing Subsystem
 SPU Scene Processing Unit
 SQA Software Quality Assurance
 SRCDR Software Requirements and Conceptual Design Review
 SRCDS Software Requirements and Conceptual Design Specification
 SRR System Requirements Review
 SRS Software Requirements Specification
 SRS System Requirement Specification
 SRT Supporting Research and Technology
 SS Seconds
 S/S Subsystem
 SSA S-Band Single Access
 SSC Science Support Center
 SSDA Sequential Similarity Detection Algorithm
 SSM Support Systems Module
 SSO Space System Operations
 SSRR Systems Software Requirements Review
 SST Synchronous System Trap
 ST Space Telescope
 STACC Standard Telemetry and Command Components
 STACC-CU STACC Central Unit
 STACC-STINT STACC Interface Unit
 STC System Test Console
 STD System Test Director
 STD Standard
 STDL System Test and Operation Language
 STDN Spaceflight Tracking and Data Network
 STEP Space Technology Engineering Program
 STINT Standard Interface for Onboard Computer
 STINT STACC Interface Unit
 STOCC Space Telescope Operations Control Center
 STOL System Test and Operations Language
 STP System Test Plan
 STR Standard S/C Telemetry Recorder
 STR Standard Tape Recorder
 STR System Test Review
 STS Space Transportation System
 STS Shuttle Transportation System
 STSOC Space Telescope Scientific Operations Center
 SU Switching Unit
 SVS Space Vehicle Specification
 S/W Software
 SWG Science Working Group
 SYCI System Corrected Images
 TA Transistor Adaptor
 TAC Telemetry and Command
 TAC TM Adaptive Capability
 TAG TM Archival Product Generation
 TAM Three Axis Magnetometer
 TAS Tape Archive and Storage
 TBA To Be Announced
 TBD To Be Determined
 TBD To Be Defined
 TBR To Be Resolved

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TBS To Be Supplied
 TBV To Be Verified
 T/C Time Code
 TCC Time Code Controller
 TCG Time Code Generator
 TC1/OSC Time Code In/Oscillator
 TCOM Army Test and Evaluation Command
 TCO/PAN Time Code Out/Panel
 TCS Thermal Control System
 TCU Time Code Unit
 T&D Test and Diagnostic
 TD Test Directives
 TDRS Tracking and Data Relay Satellite
 TDRSS Tracking and Data Relay Satellite System
 T&E Test and Evaluation
 TEP Telemetry Extraction Process
 TGS Transportable Ground Station
 TIPS TM Image Processing System
 TIROS-N Television Infrared Observing System
 TIS Technical Information Series
 TLM Telemetry
 TM Thematic Mapper
 TM Telemetry
 TMV Telemetry Volts
 TOD True-of-Date
 TP Telemetry Processor
 TPG Test Pattern Generator
 TPL Test Plan
 TR Tape Recorder
 TRB Test Review Board
 TRF Tracking and Receiving Facility
 TRK Track (HDDR)
 TRRG Tracking
 TRP Technical Recognition Program
 TRW TRW Defense and Space Systems Group
 T/S Thermal/Structural
 TSIM Test and Simulation Subsystem
 TSSC Technical Support Services Company
 TSSF Tape Staging and Storage Facility
 TTA Triangular Transition Adaptor
 TT&C Telemetry Tracking and Command
 TTL Transistor Logic Device
 TTY Teletype Operator Console
 TU45 1600 bpi Magnetic Tape Unit
 TU72 6250 bpi Magnetic Tape Unit
 TU78 6250 bpi Magnetic Tape Unit
 TV Television
 TWT Traveling Wave Tube
 TWTA Traveling Wave Tube Amplifier
 TX Transmit

 UARS Upper Atmosphere Research Satellite
 UBA Unibus Adaptor
 UBC Unit Block Controller
 UDDPM Unload DDP Module
 UDF Unit Development Folder
 UFD User File Directory
 UHF Ultra High Frequency
 UIC User Identification Code

U/L Uplink
 UNIBUS Universal Bus
 UQPSK Unbalanced Quadrature
 USART Universal Synchronous Asynchronous Receiver
 Transmitter
 USB Upper Side-Band
 USDA United States Department of Agriculture
 USGS United States Geological Survey
 UTC Universal Time Coordinated
 UTM Universal Transverse Mercator

 VA Value Analysis
 VAC Volts, Alternating Current
 VAP Verification Acceptance Program
 VAX-11/780 Virtual Address Extension DEC Model Computer 11/780
 VCO Voltage-Controlled Oscillator
 VCRI Verification Cross-Reference Index
 VDC Volts, Direct Current
 VE Value Engineering
 VHF Very High Frequency
 VHRR Very High Resolution Radiometer
 VICAR Video Image Communication and Retrieval
 VIP Virtually Interfaced Peripheral
 VM Value Management
 VMS Virtual Memory (Operating) System
 VPASS Video Processor and Sync Separator
 VP IR Video Processor/Image Recorder
 V/T Vacuum Thermal
 VT Verification Test
 VT78 Intelligent CRT Terminal
 VT100 Non-Intelligent CRT Terminal
 VTR Video Tape Recorder

 W/B Wideband
 WBM Wideband Module
 WBS Work Breakdown Structure
 WBSS Wideband Subsystem
 WBVT Wide Band Video Tape
 WBVTR Wide Band Video Tape Recorder
 WCS Writeable Control Store
 WFC Wide-Field Camera
 WO Work Order
 WPC Word Processor Center
 WPM Work Package Manager
 WRS World Reference System
 WSMR White Sands Missile Range
 WTR Western Test Range
 XMIT Transmit
 XMTR Transmitter

 Z Zulu Time (GMT)
 ZTS Zoom Transfer Scoop
 ZWC Zero Word Count

 μ Micro-
 μm Micrometer (10⁻⁶ Meter)
 μp Microprocessor
 μS Microsecond

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LAS Software Functions (Partial Listing)

BAYES	Max. Likelihood Classification	LINEREPR	Repair Bad Lines
BINARY	7 Functions: +, -, *, /, and, or, XOR	LIST	List and Histogram Image Window
CALAMP	Analyze CAL Lamp Data	LISTSTAT	List Stats File
CANAL	Canonical Analysis	LUTEDT	Edits LUT File
CHAROUT	Writes Annotation to Bit Plane	LUTLOD	Load Specified LUT from LUT Disk File
CLASSMAP	Generate Class Map Film Product	LUTSAV	Save LUT on Disk File
CLUSTER	Clustering	MASKSTAT	Statistics of Mask Image
COLGEN	Generate Pseudo Color Table	MEDFIL	Perform Median Filtering
COLSLIC	Movable Zero Band in Color LUT	MINDIST	Minimum Distance Classification
CONBCLS	Combine Class	MOSAIC	Mosaic Images
CONCAT	Concatenate Images	MSSA2P	Resample MSS A-Image to P-Image
CONTOUR	Contour Image	PARALL	Parallelepiped Classification
CONVOIVE	Convolve Image (Smoothing)	PFILM	Generate P-Type Film Product
COPY	Copy or Subset Image	POLYSITE	Polygonal Site Selections
COVAR	Covariance Matrix	PSEUDO	Load Pseudo Color Tables (LUTLOD Proc.)
CURSTRK	Figure Drawing with a Cursor (Graphics Proc.)	RADIOM	Apply RLUT to Image
DESPIKE	Remove Spikes	RECORD	Copies TV to TV (Thru LUT Optional)
DISCRIM	Discriminant Analysis	RENCLS	Rename Class
DROPCLS	Delete Class	RLUT	TM A-Priori RLUT Generation
DROPSITE	Delete Training Site	SAVIAT	Saves IAT B/W Configuration
EDGE	Extract Edges in Image	SCALE	Convert Halfword Image to BYTE Image
EDGE CORR	Register Images by Edge Correlation	SCANNER	Read Scanner/Digitizer
EDITSITE	Edit Training Site Coordinates	SCDFT	Perform Fourier Analysis of SCD FHM
FFT1	1-Dimensional Fourier Transform	SCROLL	Scroll Disk Image to IAT's
FFT1FL	1-Dimensional Fourier Transform Filter	SEGMOFF	Segment Offset Correction
FFT2	2-Dimensional Fourier Transform	SEGMREPR	Repair Image Blemish
FFT2FL	2-Dimensional Fourier Transform Filter	SETTV	Redefines IAT B/W Configuration
FIGLPEN	Figure Drawing with a Light Pen (Graphics Proc.)	SHADE	Shade Plot of Image Window
FILM	Generate Film Product	SITES	Rectangular Site Selections
FIT	Scale Image by Histogram	SPLIT	Split Screen Operation
FLICKER	Blink Mode Display	STATS	Generate Stats File (Training Site)
FROMTV	Quick Copy of IAT to Disk	STKETCH	Stretch Image Contrast
FT2PIX	Generate 2-Dimensional Fourier Display	TESTGEN	Generate Test Images
FT1PIX	Generate 1-Dimensional Fourier Display	TIEPTS	Generate Control Grid for Resampling
GCDG	Generate Geometric Correction Data	TMA2P	Resample TM A-Image to P-Image
GEOM	Perform Geometric Transformation (Rubber Sheet) for LAS	TMHIST	Histograms of TM Image for RLUT
GRAPHICS	Graphic Functions Via Console Button	TOTV	Quick Copy of 'TV-Size' Image to IAT
GREYREG	Register Images by Grey Level Correlation	TRANSDIV	Transform Divergence
GRSLIC	Movable Zero Band in LUT	UMAP	Uniformity Mapping
HINDU	Histogram Inspired Cluster	UNARY	5 Functions: +, *, NOT, EXP, LOG
HISTEQ	Histogram Equalization RLUT Generation	VEGGIN	Vegetative Index
JITTER	Analyze Jitter Effects	WTGEN	Weight Generator for FFT2FL
KARLOV	Karhunen-Loeve Transform	XFERSITE	Transfer Training Site
LINEOFF	Line Offset Correction	ZOOM	Enlarge or Reduce Image

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INVITEES

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OF POOR QUALITY

Address, Stan
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-6449

Aeppli, Ted, M-7235
GE Space Division
P.O. Box 8555
Philadelphia, PA 19101
(215) 962-3870

Alberts, Larry
ORI, Inc.
1400 Spring Street
Silver Spring, MD 20910
(301) 588-6180

Alford, Bill
Code 932
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5586

Anderson, James
Code HA20, Bldg. 1100
NASA/National Space
Technology Laboratories
NSTL Station, MS 39529
(601) 688-3830

Anuta, Paul
LARS/Purdue University
W. Lafayette, IN 47906
(317) 494-6305

Ball, Dave
CSC
8728 Colesville Road
Silver Spring, MD 20910
(301) 344-6589

Balla, John
Code 726
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-6091

Banks, Gary
Code 726
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5558

Barker, John
Code 923
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8978/8881

Beaver, Judi
ORI, Inc.
1400 Spring Street
Silver Spring, MD 20910
(301) 588-6180

Bender, Lee
U.S. Geological Survey
730 National Center
12201 Sunrise Valley Drive
Reston, VA 22092
(703) 860-6273

Bernstein, Ralph
IBM Palo Alto Scientific Center
P.O. Box 10500
1530 Page Mill Road
Palo Alto, CA 94303-0821
(415) 855-3126

Beyer, Eric, M-7235
GE Space Division
P.O. Box 8555
Philadelphia, PA 19101
(215) 962-3572

Billingsley, Fred
Code 198-231
Jet Propulsion Laboratories
Pasadena, CA 91103
FTS 792-2325

Blodget, Herb
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8997

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Bly, Beldon
CSC
8728 Colesville Rd.
Silver Spring, MD 20910
(301) 589-1545

Bracken, Peter
Code 500
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-9688

Brandshaft, Don
Hughes Aircraft Corp.
Santa Barbara Research Center
75 Coromar Drive
Santa Barbara, CA 93317
(805) 968-3511 x343

Brooks, Joan
General Electric
Space Division
4701 Forbes Boulevard
Lanham, MD 20706
(301) 459-2900, Ext. 455

Brown, Lottie
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8520

Buhler, Lynn
ORI, Inc.
1400 Spring Street
Silver Spring, MD 20910
(301) 588-6180

Busse, Jon
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-6142

Campbell, William J.
Code 744
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8116

Carr, Jim
ORI, Inc.
1400 Spring Street
Silver Spring, MD 20771
(301) 588-6180

Cicone, Richard
Environmental Research
Institute of Michigan
P.O. Box 8618
Ann Arbor, MI 48107
(313) 994-1200

Clark, Bill
CSC
8728 Colesville Road
Silver Spring, MD 20910
(301) 589-1545

Clark, Judy
ORI, Inc.
1400 Spring Street
Silver Spring, MD 20910
(301) 588-6180

Colwell, Robert
Associate Director
Space Sciences Laboratory
University of California,
Berkeley
Berkeley, CA 94720
(415) 642-2351

Connors, Kathy
ORI, Inc.
1400 Spring Street
Silver Spring, MD 20910
(301) 588-6180

Cox, Scott
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8909

Cressy, Phil
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-7658

Crone, Larry
NOAA
National Earth Satellite Service
FB No. 4- Room 0135
Washington, DC 20233

Dallam, William
General Electric
Space Division
4701 Forbes Blvd.
Lanham, MD 20706
(301) 459-2900

Dasgupta, Rangit
CSC
8728 Colesville Road
Silver Spring, MD 20910
(301) 589-1545, Ext. 722

DeGloria, Stephen
Space Sciences Laboratory
University of California, Berkeley
Berkeley, CA 94720

Dietz, Jim
Code 435.9
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-9456

Dozier, Jeffrey
Associate Professor
of Geography
University of California
Santa Barbara, CA 93106
(805) 961-2309 or 2109

Dunker, Chris
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-4931

Dykstra, John
Earth Satellite Corp.
7222 47th St.
Chevy Chase, MD 20815
(301) 652-7130

Eng, Kenneth
Systems & Applied
Sciences Corp.
5809 Annapolis Rd.
Hyattsville, MD 20784
(301) 699-5400

Engel, Jack
Hughes Aircraft Corp.
Santa Barbara Research Center
75 Coromar Drive
Santa Barbara, CA 93317
(805) 968-3511, Ext. 6145

Erickson, Jon
Earth Resources Applications
Division
Mail Code SH
NASA/Johnson Space Center
Houston, TX 77058
(FTS) 525-4017

Everett, John
Earth Satellite Corp.
7222 47th Street
Chevy Chase, MD 20815
(301) 652-7130

Fischel, David
Code 932
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-9534

Foote, Harlan

Foster, James
Code 924
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-9135

Freden, Stan
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5818

Gervin, Janette
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-7061

Gonzales, Lou
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5161

Gordon, Frederick
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8037

Goward, Samuel
NASA/Goddard Institute
for Space Studies
2880 Broadway
New York, NY 10025

Grebowsky, Gerry
Code 564.3
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-6386

Gurney, Charlotte
Systems & Applied Sciences Corp.
5809 Annapolis Road
Hyattsville, MD 20784
(301) 699-6137

Hahn, Jerald
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-6568

Haight, Steve
ORI, Inc.
1400 Spring Street
Silver Spring, MD 20910
(301) 588-6180

Hallada, Wayne

Heffner, Paul
Code 564.3
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-6263

Heinig, Joe
Code 564.3
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-7506

Hill, Charles
Test & Evaluation Group
NASA/Earth Resources Lab
NSTL Station, MS 39529
(FTS) 494-2042

Hlauka, Chris
Ames Research Center
NASA
Moffett Field, CA 94035

Horn, Tim
Code 435.9
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-7876

Horseman, Martha
Code 920
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8671

Hovis, Warren
Director, Satellite Experiment
Lab
National Oceanic & Atmospheric
Administration
National Earth Satellite Service
FB NO. 4-Room 0135
Washington, DC 20233
(301) 763-7381

Imhoff, Marc
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-7095

Irons, James
Code 923
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5240

Jackson, Michael J.
Natural Environment Research Council
Polaris House
North Star Avenue
Swindon
Wilts. SN2 1EU
UNITED KINGDOM
Telex: 444293 (ENVRE G)

Johnson, Robert
Hughes Aircraft
Space & Communications Group
El Segundo, CA

Kaufman, Lynn
ORI, Inc.
1400 Spring Street
Silver Spring, MD 20910
(301) 588-6180

Kerber, Arlene
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5355

Kieffer, Hugh
Branch of Astrogeologic Studies
U.S. Geological Survey
2255 North Gemini Drive
Flagstaff, AZ 86001
(FTS) 261-1357

Koepp-Baker, Nick M-7226
GE Space Division
P.O. Box 8555
Philadelphia, PA 19101
(215) 962-2238

Koffler, Russ
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5207

Krehbiel, John
Code 733
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5777

Krueger, Don
Code 5118
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5118

Kugelmann, Diane
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8146

Labovitz, Mark
Code 922
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5600

Lansing, Jack
Hughes Aircraft Corp.
Santa Barbara Research Center
75 Coromar Drive
Santa Barbara, CA 93317
(805) 968-3511

Latty, Rick
Code 923
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-9256

Lauer, Donald
Applications Branch
U.S. Geological Survey
EROS Data Center
Sioux Falls, SD 57198
(FTS) 784-7111

Lauritson, Levin
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5223

Linstrom, Loren
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5223

Lu, Yun-chi
CSC
8728 Colesville Rd.
Silver Spring, MD 20910
(301) 589-1545

Lyon, John
Code 932
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8744

MacDonald, Robert
Earth Resources Research
Division
NASA/Johnson Space Center
Mail Code SG
Houston, TX 77058
(713) 525-6141

Malherbe, Pete, M-7222
GE Space Division
P.O. Box 8555
Philadelphia, PA 19101
(215) 962-4699

Malila, William
Environmental Research Institute
of Michigan
P.O. Box 8618
Ann Arbor, MI 48107
(313) 994-1200

Manheimer, Harry
NASA Headquarters
600 Independence Ave., S.W.
Washington, DC
(202) 755-1201

Markham, Brian
Code 923
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5240

Martucci, Louis M.
DOE Pacific Northwest
Laboratory
Richland, WA 99352
(202) 252-2146 (temporary)

Maxwell, Marvin
Code 920
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8036

Middleton, Elizabeth
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8403

Moore, Jesse
NASA Headquarters
600 Independence Ave., S.W.
Washington, DC
(202) 755-3728

Mowle, Ed
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-6978

Mulligan, Patricia
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5515

Murphy, Bob
Code 923
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-7282

Nelson, Ross
Code 923
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-4926

Nieman, Ron
CSC
8728 Colesville Road
Silver Spring, MD 20910
(301) 589-1545

Ornisby, James
Code 924
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-6908

Oseroff, Harold
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8933

Podwysocki, Mel

Price, John
Hydrology Laboratory
Beltsville Agriculture
Research Center, West
Beltsville, MD 20705
(301) 344-3490

Prokop, Ed
Code 564.3
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-7506

Quann, John
Code 500
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8768

Ramipriyan, H. K.
Code 932
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-9406

Rango, Al
Code 924
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5480

Robinson, Jon
CSC
8728 Colesville Rd.
Silver Spring, MD 20910
(301) 589-1545

Royal, Al
General Electric Space
Division
4701 Forbes Blvd.
Lanham, MD 20705
(301) 459-2900

Salomonson, Vince
Code 920
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-6481

Schnetzler, Charles
Code 922
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5213

Schott, John
School of Photographic Arts
& Sciences
Photographic Science &
Instrumentation Division
Rochester Institute of Technology
One Lomb Memorial Drive
Rochester, NY 14623
(716) 475-2783

Sehn, George, M-7235
GE Space Division
P.O. Box 8555
Philadelphia, PA 19101
(215) 962-3852

Sheffield, Charles
Farn Satellite Corp.
7222 47th St.
Chevy Chase, MD 20815
(301) 652-7130

Short, Nick
Code 922
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-7870

Slater, Philip
Optical Sciences Center
University of Arizona
Tucson, AZ 85721
(602) 626-4242

Staskowski, Ron
Earth Satellite Corp
7222 47th St.
Chevy Chase, MD 20815
(301) 652-7130

Stauffer, Mark
Code 923
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771

Strome, W. M.
Canada Center for Remote Sensing
2464 Sheffield Road
Ottawa, Ontario, Canada K1A 0X7

Stuart, Locke
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-6987

Sudey, John
Code 716
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-4907

Thornodsgard, June
EROS Data Center
Sioux Falls, SD 57198
(605) 594-6555

Thompson, R. J.
EROS Data Center
Sioux Falls, SD 57198
(605) 594-6555

Toll, Dave
Code 923
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-9256

Townshend, J.
Natural Environment Research Council
Polaris House
North Star Avenue
Swindon
Wilts. SN 2 1EU
UNITED KINGDOM
Telex: 444293 (ENVRE G)

Ungar, Steve
NASA/Goddard Institute for
Space Studies
2880 Broadway
New York, NY 10025

Van Wie, Pete
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-7605

Waltz, Fred
EROS Data Center
Sioux Falls, SD 57198
(605) 594-6555

Warriner, Howard

Watt, Bill
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-9437

Webb, William
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8308

Weinstein, Oscar
Code 435
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8108

Welch, Jim
NASA Headquarters
Exploration Division, OSSA
600 Independence Ave., S.W.
Washington, DC 20546
(202) 755-8458

Weich, Roy
Department of Geography
University of Georgia
Athens, GA 30602
(404) 542-2856

Wescott, Tom
GE Space Division
4701 Forbes Blvd.
Lanham, MD 20706
(301) 459-2900

Whitman, Ruth
ORI, Inc.
1400 Spring Street
Silver Spring, MD 20910
(301) 588-6180

Williams, Darrel
Code 923
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-8860

Witt, Ronald
Code 902
NASA/Goddard Space
Flight Center
Greenbelt, MD 20771
(301) 344-5042

Wrigley, Robert
MS242-4
Ames Research Center
NASA
Moffett Field, CA 94035
(FTS) 448-6060

Wukelic, George

Zobrist, Albert
MS 168-514
Jet Propulsion Laboratory
California Institute of
Technology
4800 Oak Grove Drive
Pasadena, CA 91109
(FTS) 792-3237

END

DATE

FILMED

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