

EFFECTS OF DEFECTS IN COMPOSITE
STRUCTURES

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DESIGN CRITERIA

FIRST PLY FAILURE

- DESIGN ULTIMATE STRAIN (DUS) IS STRAIN FOR FIRST PLY FAILURE

STATE OF THE ART

- DUS IS FAILURE STRAIN OF LAMINATE WITH 0.25-in. HOLE

POSSIBLE

- DUS IS FAILURE STRAIN OF LAMINATE WITH LOW-ENERGY IMPACT DAMAGE
- BUFFER AND SOFTENING STRIPS USED

UPPER BOUND

- DUS IS FAILURE STRAIN OF LAMINATE
- STRESS CONCENTRATIONS DESIGNED AROUND LEAST CONSERVATIVE

PREPREG DEFECTS

HOLLOW FIBERS

EXCESSIVE VARIABILITY IN FIBER PROPERTIES

RESIN-STARVED OR FIBER-STARVED AREAS

WRINKLES, WAVINESS, MISCOLLIMATION

FOREIGN PARTICLES, CONTAMINATION

PILLS AND FUZZ BALLS

NONUNIFORM AGGLOMERATION OF HARDENER

PREPREG OUT OF SPECS.

DEFECTS IN LAMINATES

HOLLOW FIBERS
FIBER BREAKS
EXCESSIVE POROSITY, VOIDS
RESIN-RICH AND RESIN-STARVED AREAS
FIBER WAVINESS, WRINKLES, MISCOLLIMATION
FOREIGN PARTICLES, CONTAMINATION, INCLUSIONS
INCOMPLETE AND/OR VARIABLE CURE
WRONG STACKING SEQUENCE
DENTS, TOOL IMPRESSIONS, SCRATCHES

DELAMINATIONS
PLY GAPS

LAMINATE POROSITY

STUDIED EXTENSIVELY

MATRIX DOMINATED PROPERTIES DEGRADED (DELAMINATION NOT INCLUDED)

- 5% STRENGTH REDUCTION FOR 1% POROSITY
- 50% LIFE REDUCTION FOR 1% POROSITY

FIBER-DOMINATED PROPERTIES NOT AFFECTED

DELAMINATION GROWTH AFFECTED - NOT WELL DOCUMENTED

MOISTURE ABSORPTION

- EQUILIBRIUM MOISTURE LEVELS INCREASED
- AGGRAVATES THERMAL SPIKE PHENOMENON

EFFECT OF PLY GAP DEFECT

(REF. 1)

$[(0/45/90/-45)_S]_2$ LAMINATE

- 16.9% STRENGTH REDUCTION FOR GAP(S) IN 90 PLIES
- 8.7% REDUCTION FOR GAP(S) IN 0 PLIES

$[(0/45/0/-45/0)_S]_2$ LAMINATE

- 6.5% STRENGTH REDUCTION FOR GAPS IN 0 PLIES

(REF. 2)

$[(0/\pm 45/90)_S]_2$ LAMINATE

- 12.8% STRENGTH REDUCTION FOR GAPS IN OUTER 45 PLIES

$[(0/-45/0)_S]_2$ LAMINATE

- 6.2% STRENGTH REDUCTION FOR GAPS IN OUTER 45 PLIES

DEFECT CRITICALITY - BENIGN FOR DESIGN ULTIMATE STRAIN 0.7%

EFFECT OF PLY WAVINESS DEFECT

(REF. 3)

SURFACE 0 PLY WAVINESS IN $[(0/45/90/-45)_S]_2$ LAMINATE

STATIC TENSILE STRENGTH REDUCTION

- 10% FOR SLIGHT WAVINESS
- 25% FOR EXTREME WAVINESS

FATIGUE LIFE REDUCTION

- AT LEAST A FACTOR OF 10
- CONSISTENT WITH STATIC STRENGTH REDUCTION

DEFECT CANNOT BE FOUND BY STANDARD NDE

STRENGTH LOSS CAN BE PREDICTED BY ASSUMING LOSS OF LOAD CARRYING CAPACITY DUE TO THE WAVINESS

DEFECT CRITICALITY - INSUFFICIENT DATA FOR ACCURATE ASSESSMENT

- SHOULD BE BENIGN FOR DESIGN ULTIMATE STRAINS 0.7%

MACHINING DEFECTS

EDGE DELAMINATIONS	EDGE NOTCHES AND SURFACE NOTCHES
OVERSIZE HOLES	HEAT-DAMAGED MACHINED EDGES
UNDERSIZE HOLES	FIBER BREAK-OUT ON HOLE EXIT SIDE
TILTED HOLES	OUT-OF-ROUND HOLES
TILTED COUNTERSINKS	IMPROPER DEPTH OF COUNTERSINKS

DENTS, FIBER BREAKING FROM IMPACT

TEAROUT OR PULL-THROUGH IN COUNTERSINKS

EFFECT OF SURFACE NOTCHES

EXPERIMENTAL DATA

STATIC STRENGTH REDUCED UP TO 50%

LOCAL DELAMINATION AT NOTCH

FATIGUE LOADING REDUCES STRESS CONCENTRATION

RESIDUAL STRENGTH HIGHER THAN STATIC STRENGTH

DATA AVAILABLE FOR VARIOUS STACKING SEQUENCES

ANALYSIS

ECCENTRIC BEAM MODEL PREDICTS STRENGTH REDUCTION

STRENGTH REDUCTION IS SMALL FOR SIZES EXPECTED
IN SERVICE

DEFECT CRITICALITY

NOT CRITICAL FOR $DUS < 0.7$

BOLTED ASSEMBLY DEFECTS

OVERTORQUED FASTENERS IMPROPER FASTENER SEATING
MISSING FASTENERS FASTENER INSTALLATION DAMAGE
OVERSIZED AND UNDERSIZED FASTENER

BONDING DEFECTS

ADHESIVE POROSITY
MISCURE
ADHESIVE-STARVED AREAS
IMPROPER SURFACE PREPARATION

REFERENCES

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3. Ryder, J. T.: Effect of Load History on Fatigue Life. AFWAL-TR-80-4044, U. S. Air Force, June 1980.