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THE UNIVERSITY OF TEXAS AT AUSTIN Institute For Geophysics EARTH STRAIN MEASUREMENTS WITH THE TRANSPORTABLE LASER RANGING SYSTEM: FIELD TECHNIQUES AND PLANNING

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ABSTRACT

We have conducted a feasibility study to examine the potential of the Transportable Laser Ranging System (TLRS) for monitoring the ground deformation around satellite ranging stations and other geodetic control points. Emphasis has been placed on testing the usefulness of the relative lateration technique. The temporal variation of the ratio of the length of each survey line to the mean length of all survey lines in a given area is directly related to the mean (h)ar strain rate for the area. The data from a series of experimental measurements taken over the Los Angeles basin from a TLRS station at Mt. Wilson show that such ratios can be determined to an accuracy of one part in 10^7 with a measurement program lasting for three days and without using any corrections for variations in atmospheric conditions. A numerical experiment using a set of hypothetical data indicates that reasonable estimates of the present shear strain rate and the direction of the principal axes in southern California can be deduced from such measurements over an interval of one to two years. Thus, the relative lateration from the TLRS appears to be a very economical way to monitor ground deformations, although there has been no opporunity yet to measure the actual ground strain by reoccupying the Mt. Wilson site.

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I. INTRODUCTION

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With the recent development in ground-to-satellite laser ranging and Very Long Baseline Interferometry (VLBI) techniques, it is now possible to measure precisely distances between locations separated by several hundreds to thousands of kilometers. This makes it possible to monitor relative movements of globally distributed points on the earth for geodynamic studies. However, one question that must be answered is how representative each of these positions thus occupied is for the region in which it is located. If some of these locations are experiencing localized movements which are not representative of the region, the global measurement would give erroneous results. An answer to this questions can be found by measuring regional deformations around each location.

A conventional method for determining regional deformations is to perform repeated survey using an electro-optical distance measuring (EDM) device (e.g. Savage et al. [1981]). However, such surveys are expensive, and are rather limited in range. We, therefore, have looked for a better alternative. The development of the Transportable Laser Ranging System (TLRS) for ground-to-LAGEOS (Laser Geodynamics Earth Orbiting Satellite) ranging [Silverberg and Byrd, 1981] has given us an opportunity to test such an alternative. Because of its high sensitivity, being capable of detecting single photon returns, the TLRS can measure distances to small targets (retro-reflectors) at any visible points much beyond the normal ranges of other EDM devices. Thus, this system may provide economical measurements of strain fields in areas more than 200 km in diameter. If successful, such measurements will be valuable not only in the immediate neighborhood of satellite ranging stations, but also in understanding the dynamic behavior of both plate boundaries and areas internal to plates.

We have conducted a limited feasibility study to examine this potential. Although the TLRS is a powerful system, it also has certain limitations when used for a ground-to-ground ranging. The most important is the uncertainty of measurement results due to variability in atmospheric conditions. To bypass this problem and avoid the expense of flying an aircraft to monitor the atmospheric conditions along the path of the laser beam, we have examined the use of the relative lateration, or the ratio method, which was used earlier by Carter and Vincenty [1978] in an experimental survey around the McDonald Observatory.

We originally planned repeated field experiments at several sites in the western United States. However, because of many scheduling conflicts and delays associated with the overall TLRS-LAGEOS ranging experiments, the only field experiment we could perform during the current contract was a four-day measurement at Mt. Wilson over the Los Angeles basin in January, 1981. We have been unable to reoccupy this site for an actual strain measurement. The present study, however, has given us some very encouraging results. Even with no atmospheric correction at all, the range ratios could be determined to an accuracy of one part in 10⁷. This is sufficient for an order-of-magnitude estimate of incremental shear strain in the southern California region if two measurements separated by one to two years are available. Higher accuracies would be attainable with repeated measurements.

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In this report, we first describe the advantages and problems of ground-to-ground ranging by a TLRS, leading to the use of relative lateration, or range-ratio method, and its relationship to the regional strain (Section II). Then, we present the data and analysis of the Mt. Wilson experiment (Section III). This is followed by a short treatment of regional strain determination using hypothetical data (Section IV). Finally, we present the conclusions from this feasibility study and offer some recommendations. Some pertinent data are presented in the Appendix.

II. TLRS GROUND-TO-GROUND RANGING

Advantanges and Problems

The TLRS is a highly mobile satellite laser ranging system designed to perform ground-to-LAGEOS range measurements. It is also highly sensitive, being capable of determining the range to a LAGEOS satellite with return signals as low as one photelectron every 20 to 50 laser shots [Silverberg <u>et al.</u> 1982]. Used as a ground-to-ground ranging device, it can measure the distance to any single 1 inch (25 mm) corner reflector within sight at very low laser power level. The measureable range is limited only by the curvature of the earth. The required power level is so low that, unlike some systems used for similar measurements, the laser beam can be maintained many orders of magnitude below the eye-damage threshold.

The practical precision of the TLRS range measurements is limited to about 1.5 cm for a one minute average, which is somewhat worse than those of conventional EDM devices using modulated laser beams. However, the long range capability of the TLRS reduces the relative error to well within the limits of interest in conventional surveys. The TLRS has an automatic pointing system, an automatic calibration system and other features which lend themselves to providing many horizontal (ground-toground) line measurements on an operational basis. Thus it will be a good device to use if the data it provides is sufficient to determine the regional deformation at a high-enough accuracy.

The most serious problem in using the TLRS for ground-to-ground ranging is the atmospheric effect. The temperature, pressure, and to a lesser degree water vapor influence the index of refraction of air, and thus the speed of a laser beam through the atmosphere. To obtain the absolute distance between two points from a time of flight measurement through air, one must make corrections for these atmospheric variables.

Estimates of these atmospheric variables along the beam path may be made based on measurements at the two end points. This, however, is unsatisfactory for long lines. A more precise way is to measure directly the atmospheric condition along the beam path by flying an aircraft during the ranging. This, though done in practice, is a costly operation. A third alternative is to use more than one wavelength for ranging. Using the dispersive characteristics of light in air, one can correct for the atmospheric effects [Huggett, et al. 1977].

The present TLRS operates in a single color. Flying an aircraft, we judged, is too costly for repeated measurements in many directions. Thus we had to look for another alternative.

Relative Lateration

One way to improve the accuracy of range measurements without relying on expensive in-flight measurement of atmospheric conditions is to use a relative lateration technique, or the "ratio method" (Robertson, 1972). Instead of attempting to measure the absolute length of each survey line to high accuracy, this technique determines only the ratios of distances. This method is based on a supposition that the temporal changes of atmospheric conditions along several survey lines within a given region are similar to each other. Therefore, even when the time of flight of a laser beam in each line fluctuates with changing atmospheric conditions, the ratios of the times of flight along different survey lines tend to vary little with time.

Carter and Vincenty [1978] used this method in an experimental EDM survey around the McDonald Observatory in 1977. They obtained sets of measurements, one month apart, consistent to one to two parts in 10^7 . They have just repeated this experiment and the data is now being analyzed. Since the results of Carter and Vincenty appear to be quite promising, we have decided to try the same for our TLRS measurements.

Relationship Between Relative Lateration and Strain

Unlike absolute measurements of distances, the relative distance measurements repeated after a certain time period will not give all of the components of deformation, or incremental strain, for the time period unless at least one survey line is measured absolutely. However, a clear relationship exists between the changes of relative distances and incremental shear strain.

Let us consider n survey lines radiating from a central station. In the present case, the TLRS is located at the central station and a retroreflector is located at the end of each radiating line. Assume that all lines lie in a horizontal plane, neglecting both the curvature of the earth's surface and topographic height differences. Choosing a coordinate system with the origin at the central station, positive x towards east and positive y towards north, the original length of line i to the reflector at coordinates (x_i, y_i) at the time of the initial survey is given by

$$s_{i} = (x_{i}^{2} + y_{i}^{2})^{\frac{1}{2}}$$
(1)

Now assume that between the initial survey and a subsequent survey the entire area of the survey undergoes a uniform deformation represented by incremental strain components ε_{xx} , ε_{xy} and ε_{yy} . Then, the line length becomes

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$$\mathbf{s_{i}}' = [(\mathbf{x_{i}} + \varepsilon_{\mathbf{xx}}\mathbf{x_{i}} + \varepsilon_{\mathbf{xy}}\mathbf{y_{i}})^{2} + (\mathbf{y_{i}} + \varepsilon_{\mathbf{xy}}\mathbf{x_{i}} + \varepsilon_{\mathbf{yy}}\mathbf{y_{i}})^{2}]^{\frac{1}{2}}$$

$$= [\mathbf{x_{i}}^{2} + \mathbf{y_{i}}^{2} + 2\varepsilon_{\mathbf{xx}}\mathbf{x_{i}}^{2} + 4\varepsilon_{\mathbf{xy}}\mathbf{x_{i}}\mathbf{y_{i}} + 2\varepsilon_{\mathbf{yy}}\mathbf{y_{i}}^{2}]^{\frac{1}{2}}$$

$$= \mathbf{s_{i}}[1 + 2\varepsilon_{\mathbf{xx}}\sin^{2}\alpha_{i} + 4\varepsilon_{\mathbf{xy}}\sin\alpha_{i}\cos\alpha_{i} + 2\varepsilon_{\mathbf{yy}}\cos^{2}\alpha_{i}]^{\frac{1}{2}}$$

$$= \mathbf{s_{i}}[1 + \varepsilon_{\mathbf{xx}}\sin^{2}\alpha_{i} + \varepsilon_{\mathbf{xy}}\sin\alpha_{i}\cos\alpha_{i} + 2\varepsilon_{\mathbf{yy}}\cos^{2}\alpha_{i}]^{\frac{1}{2}}$$
(2)

where $\alpha_i = \tan^{-1}(x_i/y_i)$ is the azimuth of the line i measured clockwise from north, and the higher order terms in strain have been neglected. Then, the range increment δ_i is given by

$$\delta_{i} = s_{i}' - s_{i} = s_{i} [\varepsilon_{xx} \sin^{2} \alpha_{i} + \varepsilon_{xy} \sin 2\alpha_{i} + \varepsilon_{yy} \cos^{2} \alpha_{i}]$$
(3)

Next define original mean range and range ratios to the mean, respective, as

$$\overline{s} = \sum_{i=1}^{n} s_i / n$$
(4)

and

$$r_{i} = s_{i}/\overline{s}$$
(5)

Then, the subsequent mean range and range ratios are

$$\overline{\mathbf{s}}' = \sum_{i=1}^{n} \mathbf{s}_{i}'/n = \overline{\mathbf{s}} + \sum_{i=1}^{n} \delta_{i}/n$$
(6)

and

$$r_{i}' = s_{i}'/\overline{s}' = (s_{i} + \delta_{i})/(\overline{s} + \frac{n}{i \equiv 1} \delta_{i}/n)$$
$$= r_{i}(1 + \delta_{i}/s_{i} - \frac{n}{i \equiv 1} \delta_{i}/n\overline{s})$$
(7)

where the higher order terms are again neglected. The increment of the range ratio is, therefore,

$$\mathbf{r}_{i}' - \mathbf{r}_{i} = \mathbf{r}_{i} \left(\delta_{i} / \mathbf{s}_{i} - \frac{n}{i=1} \delta_{i} / n \mathbf{s} \right)$$
(8)

Then, range ratio increment normalized by the original range ratio is given by

$$\gamma_{i} = (r_{i}' - r_{i})/r_{i} = \delta_{i}/s_{i} - \sum_{i=1}^{n} \delta_{i}/ns$$
(9)

Substituting eq. (3) into eq. (9), and using (5), we obtain

$$\gamma_{i} = [\sin^{2} \alpha_{i} - \sum_{i=1}^{n} (r_{i} \sin^{2} \alpha_{i})/n] \epsilon_{yy}$$

$$+ [\sin^{2} \alpha_{i} - \sum_{i=1}^{n} (r_{i} \sin^{2} \alpha_{i})/n] \epsilon_{xy}$$

$$+ [\cos^{2} \alpha_{i} - \sum_{i=1}^{n} (r_{i} \cos^{2} \alpha_{i})/n] \epsilon_{yy} \qquad (10)$$

Equation (10) may give one an impression that a set of measurements of the normalized range ratio increments $\boldsymbol{\gamma}_i$ would give the incremental strain components ε_{XX} , ε_{XY} and ε_{YY} . However, this impression is incorrect because the coefficients of ε_{xx} and ε_{yy} are not independent of each other, as their sum vanishes, and therefore ε_{XX} and ε_{VY} cannot be determined uniquely.

Now let

$$\Theta = \varepsilon_{\mathbf{x}\mathbf{x}} + \varepsilon_{\mathbf{y}\mathbf{y}}$$
(11)

and

$$\Psi = \varepsilon_{\mathbf{X}\mathbf{X}} - \varepsilon_{\mathbf{Y}\mathbf{Y}} \tag{12}$$

Then,

 $\varepsilon_{\mathbf{x}\mathbf{x}} = \frac{1}{2} (\Theta + \Psi)$ (13)

and

$$\varepsilon_{yy} = \frac{1}{2} (\Theta - \Psi) \tag{14}$$

Substituting (13) and (14) into (10), we obtain

$$\gamma_{i} = [\sin 2\alpha_{i} - \sum_{i=1}^{n} (r_{i} \sin 2\alpha_{i})/n] \varepsilon_{xy}$$
$$- \frac{1}{2} [\cos 2\alpha_{i} - \sum_{i=1}^{n} (r_{i} \cos 2\alpha_{i})/n] \Psi$$
(15)

The coefficients of $\varepsilon_{\mathbf{X}\mathbf{Y}}$ and Ψ are known quantities for the initial setup of the survey lines. Thus, for a set of measurements of the normalized range ratio increments γ_i , the incremental shear strain components ε_{xy} and Ψ can be determined by a least-square inversion of eq. (15).

Finally, the maximum incremental shear strain S and the direction of the principal strain axes β are given by

$$s = [(2\varepsilon_{xy})^{2} + \Psi^{2}]^{\frac{1}{2}}$$
(16)
$$\beta = \frac{1}{2} \tan^{-1} (2\varepsilon_{xy}/\Psi)$$
(17)

and

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The dilation Θ of eq. (11) disappears from eq. (15), and thus cannot be determined. This is expected because any uniform compression or expansion of the entire area causes no change in range ratios.

The treatment above assumes uniform deformation of the entire region. If for some reason, such as the existence of active faults within the area, the regional deformation is not uniform, large residuals will show up in the least-square inversion of eq. (15). Thus, any residuals significantly larger than the measurement errors will indicate heterogeneous strain.

The remaining question is how accurately we can estimate the normalized range ratio increments γ_i . Since the measurements are done in terms of time of flight of light beams, the uncertainty in speed of light is the determining factor. The average speed of light, c_i , between the central station and a reflector i may be expressed as the sum of four components:

$$c_{i} = c_{o} + \ell_{i} + w_{c} + w_{i} \tag{18}$$

where c_0 is the speed of light in standard air, which is constant for all survey lines at all times; l_i is the correction attributable to the reflector location, which is time invariant for a given reflector; w_c is a component of correction attributable to weather common to all reflectors at a given time; and w_i is the residual weather correction. The line length s_i is given in terms of round-trip time of flight, t_i , as

$$s_{i} = \frac{1}{2}(c_{o} + l_{i} + w_{c} + w_{i})t_{i}$$
(19)

the mean range as

$$\overline{s} = \frac{1}{2} \left(c_0 \overline{t} + \sum_{i=1}^{n} \ell_i t_i / n + w_c \overline{t} + \sum_{i=1}^{n} w_i t_i / n \right)$$
(20)

where $\bar{t} = \Sigma t_i/n$ is the mean time of flight, and the range ratio to the mean as i=1

$$r_{i} = u_{i} [1 + (l_{i} - \sum_{i=1}^{n} l_{i}t_{i}/n\bar{t} + w_{i} - \sum_{i=1}^{n} w_{i}t_{i}/n\bar{t})/c_{0}]$$
(21)

where $u_i = t_i/t$ is the time-of-flight ratio to the mean, and the higher-order terms have been neglected. Finally, the normalized range ratio increment 1s given as

$$\gamma_{i} = \eta_{i} + [(w_{i}' - w_{i}) - \sum_{i=1}^{n} (w_{i}' - w_{i})t_{i}/n\bar{t}]/c_{o}$$
(22)

where $n_i = (u_i' - u_i)/u_i$ is the normalized time-of-flight ratio increment and quantitles with primes designate those at subsequent measurement as before. The higher-order terms are again Beglected. Note that the common weather component, w, is eliminated by taking the range ratio (21), and the location specific components, k_i 's, are eliminated by normalization (22), leaving only the residual weather components w_i 's.

The normalized time-of-flight ratio increment, η_i , thus approximates the range ratio increments, γ_i , with a small error due to residual weather term. The latter is not location specific, and is not common to all lines at a given time. If this term is sufficiently small, then we can substitute η_i for γ_i in calculating the shear strain increment using (15).

III. MT. WILSON EXPERIMENT

Field Experiment

At the request of the NASA Crustal Dynamics Project, the TLRS team from the McDonald Observatory of the University of Texas, led by Dr. Eric Silverberg, deployed the TLRS at Mt. Wilson, California, in January of 1981. At the same time, two of us (H.J.D. and T.C.) scouted the surrounding area for suitable target sites and selected the reflector locations. Then, a field party from the National Geodetic Survey (NGS), which was dispatched at the request of NASA to help us, deployed retroreflectors at the chosen sites. The survey lines selected for the site are shown in Figure 1. Table 1 lists the nominal coordinates of the base station (TLRS 1100) at Mt. Wilson and of the end points of the lines, where the retroreflectors were installed. Also listed in Table 1 are the approximate look angles from Mt. Wilson and ranges as computed from the indicated coordinates using the IAG standard ellipsoid Geodetic Reference System 1967.

Each reflector except the one at Cahuenga was a metal box containing an array of three $1\frac{1}{2}$ inch (38mm) corner cubes, supplied by the NGS. The box was mounted on a tripod and placed directly over the station mark using an optical blumb bob. This elaborate configuration made it necessary to guard the reflector continuously for the entire duration of the experiment. The reflector used at Cahuenga was designed by one of us (T_*C_*) for unmanned operation. It contained a single 1 inch (25mm) corner childs and was fastened to an outcrop with anchor bolts at a site off the station mark, thus concealed from public view.

The reference point of the TLRS, from which the raw time-of-flight measurements were made, was slightly offset from the Mt. Wilson station mark given in Table 1. The measured coordinates of the station mark relative to the TLRS were:

> x = -1.4873 m (west) y = 0.5093 m (north) z = -3.3709 m (below)

The resulting corrections, to be applied to the observed quantities to reduce them to the reference mark, are listed in Table 2. The corrections can be applied at any stage of data reduction.

After the initial setup, which began on January 9, 1981, the horizontal ranging data were collected over the four-day interval January 23 through 26, 1981, in cooperation with the NOAA National Geodetic Survey. Each of the reflector sites except Cahuenga was manned continuously during the entire experiment to record the temperature, pressure and relative humidity at the site at about 30 minute intervals. The details of the data acquisition are given in Silverberg et al. [1982].



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Station	Longitude	Latitude	Elevatio M	n Look Azimuth	Angle Altitude	Range m	Note
Mt. Wilson	241° 56' 17.85"	34° 13' 21.58"	1722.0	<u></u>	ی هم می برد می ما ای بی بی بی مرد برد م	ننگ ہیں جو جو نابہ جن پنیا ا	1
Castro	241° 12' 55.40"	34 95 08.57	858.9	257.366	-1.030	68393	2
Cahuenaa	241° 40' 29.81"	34°08'13.00"	554.1	248.687	-2.681	26104	3
San Pedro	241° 39' 55.73"	33' 44' 40.88"	447.1	205.506	-1.508	58729	4
Nique1	242° 16' 00.18"	33' 30' 44.03"	288.0	158.814	-1.353	84459	
Santiago	242 28'00.10"	33° 42' 37.89"	1733.0	139.168	-0.329	74933	
San Juan	242 15 45.99"	33° 54′ 49.47 "	543.0	138.751	-1.688	45536	
1 New marke 2 Solitice	r 9.408m from M Canuon B2 Aux.	1. Wilson E10	 A @345°17 4.107m EN	E of Cas		43030	

Table 1. Stations Used in Mt. Wilson Line Survey

2 Solitice Canyon B2 Aux. 1, which is 14.107m ENE of Castro 1898 3 Reference mark #3 of Cahuenga #2, 13.329m @257°47′ from Cahuenga #2 4 L7 Ecc. San Pedro Hills, which is 12.576m @318°57′ from San Pedro #3

	Round-Trip		Range	Ratio*
Station	Time of Flight	Range	(1) [–]	(2)
	ns	m	p	pm
Castro	-9.344	-1.4003	-25.08	-26,78
Cahuenga	-9.054	-1.3568	-23.35	-23.51
San Pedro	-1.798	-0.2694	-5.91	-7.93
Niguel	6.222	0.9325	13.61	9.93
Santiago	8.931	1.3385	20.64	17.07
San Juan	8.432	1.2636	20.09	17.64

Table 2. Corrections to be Applied to Observations to Reduce to Mi. Wilson Gound Marker

* (1) Ratio to mean range

(2) Same but excluding Cahuenga and Niguel

The Data

The raw field data were initially processed at the University of Texas at Austin by the McDonald Observatory group. As described in detail by Silverberg et al. [1982], the processing of the raw data involved accumulation of individual photon returns into 200 psec bins, smoothing of the coadded returns by three-bin (600 psec) running averages, cross-corretation with a reference standard to eliminate longterm drift in the calibration constants, adjustments to account for certain measurement irregularities, and removal of a 86.8 nsec constant calibration correction.

The calibrated round-trip time-of-flight data, shown in Figure 2 and listed in Table Al in the Appendix, have not been corrected for the offset of the TLRS from the ground marker (Table 2). The data for Cahuenga were not used for the analysis because of certain processing difficulties encountered for the data for this station.

The data gap during the second day of observation was due to an interruption in data acquisition caused by rain which accompanied the passage of a cold front. The meteorological data taken at Mt. Wilson site and other stations are shown in Figures Al through A3, and are listed in Table A2 in the Appendix.

As expected, the raw time-of-flight data show large fluctuations, which are only partially correlated with the meteorological data. The relative RMS deviations of the time-of-flight data (Table 3, colume 3) range from 1.53 ppm for Niguel, which was surveyed only after the passage of the cold front, to 3.78 ppm for San Juan, which was the shortest line. The weighted average for all lines is 2.84 ppm.

Range Ratios to a Single Reference Line

Silverberg <u>et al</u>. [1982] calculated the time-of-flight ratios and atmosphere-corrected range ratios to a reference line following the procedure used by Carter and Vincenty [1978]. The reference line they chose was a smoothed curve (a cubic spline) through the Santiago data. Their results (Table 3, column 5) show relative RMS deviations of timeof-flight ratios ranging from 0.4 ppm for Niguel to 1.6 ppm for San Juan. The weighted average for all lines is 1.0 ppm, which is about a factor of three improvement from the fluctuation of the time-of-flight data.

Their results for the range ratios with atmospheric corrections based on end-point meteorological data did not fare as well. In fact the relative RMS deviations increased typically about 40% from those of uncorrected time-of-flight ratios [Silverberg et al. 1982].

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Fig. 2. Round-trip time of flight from Mt. Wilson. The data are not corrected for the marker offset.

The main reason for the poor performance of atmosphere-corrected values is the difficulty of making proper atmospheric corrections. A comparison of the variations of the group index of refraction calculated from the temperature and pressure at end points (Figure 3; also listed in Table A3 in the Appendix) with the time-of-flight variations (Figure 2) clearly shows that long-term variations are fairly well matched but shorter diurnal fluctuations are larger for the index of refractions than for the times-of-flight. Thus the index-of-refraction correction per Carter and Vincenty [1978] over-compensates for diurnal variations.

Time-of-Flight Ratios to the Mean

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In order to be consistent with the range-ratio/strain relationship of the preceding section, we calculated the time-of-flight ratios to the mean. Since the time-of-flight measurements to all targets were not made exactly simultaneously, the data were linearly interpolated before the mean time-of-flight for a given time was calculated. (Higher order interpolations or a spline approximation might be better, but we judged the difference would be small.) Also, since we had data for Niguel only during the last half of the experiment, this station was excluded from the mean time-of-flight calculation.

The resulting time-of-flight ratios to the mean (Figure 4; also listed in Table A4 in the Appendix) show a further improvement in the fluctuations of the results. The relative RMS deviations (Table 3, column 5) now range from 0.36 ppm for Niguel to 1.24 ppm for San Juan, with the weighted average of 0.71 ppm for all lines, a factor of four improvement from the raw time-of-flight data.

An Alternative Atmospheric Correction

As stated earlier, the short-term, diurnal fluctuations in the index of refraction at end points exceed the observed fluctuations in the time-of-flight values. This is probably due to the larger fluctuation of the atmospheric temperature near the ground than those in most of the intervening air mass; a result of the base station and most of the target stations being located well above the intervening terrain. In this situation, a standard correction procedure like that of Carter and Vincenty [1978] is not really applicable, and some alternate procedures are needed.

An experimental procedure we tried was to estimate the average temperature of the air mass by low-pass filtering the mean of the temperatures measured at the end points. The filter we used was a simple one of adding all previous temperature readings each weighted by a factor proportional to a megative exponential of the elapsed time. After



Fgi. 3. Group index of refraction computed from atmospheric data at end points.

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Fig. 4. Time-of-flight ratios to the mean.

trials with such filters of several different time constants, a time constant of 12 hours was found to give the best result. The resulting relative RMS deviations of the ranges thus corrected for atmospheric conditions (Table 3, column 4) range from 0.96 ppm for Castro to 2.05 ppm for San Juan with the weighted average of 1.38 ppm for all lines. This is about a factor of two improvement from the raw timeof-flight data. However, the range ratios calculated from these corrected ranges do not show any significant improvement over those of the uncorrected ratios. The relative RMS deviations of the corrected range ratios (Table 3, last column) range from 0.52 ppm for Niguel and Santiago to 1.21 ppm for San Juan, with the weighted average of 0.72 ppm for all lines.

A comparison of the relative RMS deviations of various quantities in Table 3 reveals that the best result is obtained for the uncorrected time-of-flight ratios to the mean. The atmospheric corrections did not improve the RMS deviations at all when ratios were taken.

A Test for Systematic Error Due to Atmospheric Conditions

The reliability of the relative lateration depends on the validity of the assumption that the temporal changes of atmospheric conditions are similar for all survey lines in the area so that their effects cancel out when ratios are taken. If this assumption is incorrect, a systematic error due to varying atmospheric conditions is introduced into the measured time-of-flight ratios. The greatly different atmospheric conditions before and after the passage of a cold front during the experiment gave us an opportunity to test this assumption.

The test we performed is the likelihood ratio test. We divided the time-of-flight ratios of Table A4 for each line into two subsets, the first half and the last half, of equal size (the last half was one greater than the first half if the total number was odd). If the systematic error due to atmospheric conditions is significantly large, the mean ratio, μ_1 , for the first subset will be significantly different from that, μ_2 , for the second subset. Setting up a null hypothesis $H_0:\mu_1 = \mu_2$, if it is true, then the likelihood ratio statistic

$$t = [n_1 n_2 / (n_1 + n_2)]^{\frac{1}{2}} (\mu_1 - \mu_2) / [(n_1 \sigma_1^2 + n_2 \sigma_2^2) / (n_1 + n_2 - 2)]^{\frac{1}{2}}$$
(23)

has a t distribution with $n_1+n_2 - 2$ degrees of freedom, where n_1 and n_2 are the sample sizes of the two subsets and μ_1, μ_2, σ_1^2 and σ_2^2 are used to designate the sample means and the sample variances of the first and the second subsets, respectively, for convenience.

At 90% significance level, the t distribution has values of 1.69 for 34 degrees of freedom and 1.80 for 11 degrees of freedom, while the t values computed from the data, Table 4, are much smaller. Therefore, the null hypothesis cannot be rejected at this level of significance.

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Stat., ion	Number of Data Points	Uncorrected Time of Flight	Correcteo Range (a)	l Uncorrected T-o-F Ratio to Santiago(b)	Uncorrected T-o-F Rotio to the mean	Corrected Range Ratio to the mean
Castro	36	2.68	0.96	0.7	0.56	0.54
San Pedro	24	2.91	1.12	0.8	0.52	0.54
Niquel	13	1.53	1.81	0.4	0.36	0.52
Santiago	36	2.45	1.03	(0.2)	0.54	0,52
San Juan	27	3.78	2.05	1.6	1.24	1.21
A11	136	2.84	1.38	1.0	0.71	0.72

Table 3. Comparison of Relative RMS Deviations in ppm

(a) Corrected by Using 12-hour low-pass filtered temperature
(b) From Silverberg et al. [1982]. The deviation for Santiago is from the smoothed curve, and is not included in calculating the average for all stations.

Claiton		Subset	1		Subset	2	Degree	
5101100	1	μ	σ ₁	ⁿ 2	μ2	σ ₂	Freedom	t
Castro	18	1.104914282	0.000000513	18	1.104914438	0.000000694	34	0.745
San Pedro	12	0.948870911	0.000000356	12	0.948870818	0.000000598	22	0.445
Nique1	6	1.364486713	0.000000572	7	1.364486460	0.00000365	11	0.889
Sanilago	18	1.210566249	0.00000774	18	1.210566130	0.000000515	34	0.531
San Juan	13	0.735648159	0.00000830	14	0.735648288	0.000000983	25	0.353

Table 4. Likelihood Ratio Test for Non-equality of Means

In other words, no significant difference is found between the mean time-of-flight ratios in the first and second halves of the experiment for any of the lines surveyed.

Results

9 9 Since there is no evidence for systematic errors caused by atmospheric conditions, the most likely estimates of the mean time-of-flight ratios and their variances (and standard deviations) can be calculated from the entire data set. The results are shown in Table 5. Also listed in this table are the mean time-of-flight and the mean distances. The latter were calculated using atmospheric corrections based on the low-pass filtered temperatures described earlier and pressures interpolated to the average height of the beam from the end-point measurements (extrapolation in case of Santiago because the average height of the beam was lower than either end point). A group index of refraction of n =1.00028975 at the wavelength of 0.5320 μ m, calculated from the formula given in American Institute of Physics Handbook [1972, p. 6-111] for standard dry air with 0.03% carbon dioxide at 15°C and 760 mm Hg, is used. No other corrections have been applied to the calculated distances; thus they are subject to minor systematic errors.

The estimated relative standard deviations of the mean time-offlight ratios are approximately 1×10^{-7} except for San Juan, which is the shortest line. In comparison, Savage and Prescott [1973] estimate that the standard deviation of their Geodolite measurements of distances are 3 and 8 mm for lengths of 1 and 37 km, respectively. Thus, the precision of the present time-of-flight ratios is at least a factor of two better than that of their distance measurements. Furthermore, their distances had to be corrected for temperature and humidity readings made with an aircraft flying along the line of sight, while the present time-c[-flight ratios required no atmospheric correction at all.

Multiwavelength measurements of distances are definitely better than the above two in terms of relative accuracy. Huggett and Slater [1975] and Slater and Huggett [1976] show the standard deviation of individual distance measurements to be less than 1×10^{-7} on a 10.1 km line. By taking the mean of many measurements, which is practical in this case, the accuracy can be improved further. The ranges attainable with the multiwavelength system, however, are quite limited compared with the TLRS measurements.

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Station	Mean Time of Flight ns	Mean Distance m	Mean T-o-F Ratio	S.D, of Mean T-o-F Ratio	Relative S.D. ppm
Castro	456358.75	60308.52	1.10488758	9.00000010	0.09
San Pedro	391914.94	50730.08	0.94886293	0.00000010	0.11
Niguel	563587.77	04456.72	1.36449651	0.00000014	0.10
Santiago	500014.83	74931.62	1.21058326	0.00000011	0.09
San Juan	303856.63	45535.07	0.73566587	0.00000018	0.24

Table 5. Mean Time of Flight, Distance and Ratios*

* These results have been corrected for the TLRS/ground-marker offset.

Table 6. Increment in Normalized Ratio Due to Hypothetical Strain Increment and Rounded Values for Testing

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Station	Normalized Ratio Increment ppm	Rounded to Ø.1 ppm
Castro San Pedro Niguel Santiago San Juan	0.052 -0.158 -0.030 0.068 0.070	0.1 -0.2 0.0 0.1 0.1

V.

IV. SHEAR STRAIN DETERMINATION USING HYPOTHETICAL DATA

We have been unable to reoccupy the Mt. Wilson site for a repeat measurement, which would allow a testing of the ratio method for a shear strain determination in the region. This section, therefore, describes an exercise we have conducted to see how well we can determine the regional shear-strain increment using a set of hypothetical data.

We assume a hypothetical strain increment described by

ε	$= 0.1 \times 10^{-6}$:	maximum extension
ε_2	$= -0.2 \times 10^{-6}$:	maximum compression
ε	$-\varepsilon_2 = 0.3 \times 10^{-6}$;	maximum shear
β	= 110	:	azimuth of maximum positive principal axis (extension) measured clockwise from north

This strain increment is approximately the annual strain increment in southern California observed by Savage et al. [1981]. The resulting increments in normalized range ratios for the five survey lines used in the Mt. Wilson experiment are listed in the center column of Table 6. Since we will not be able to measure these ratio increments at this accuracy, we use the values rounded to 1×10^{-7} , as given in the rightmost column of Table 6.

Substituting these rounded ratio increments into eq. (15), and inverting it in a least-squares sense, we obtain the following results:

$$\varepsilon_1 - \varepsilon_2 = 0.40 \times 10^{-6}$$

$$\beta = 109.5^{\circ}$$

The result describes the original hypothetical shear-strain increment reasonably well. A trial with a rounding to 1×10^{-8} results in almost complete duplication of the hypothetical strain increment.

The likelihood ratio test of the preceding section can be used to estimate the required number of measurements to achieve a given level of accuracy at a given confidence level. We use the standard deviation of individual range ratio measurements of 5 x 10⁻⁷ as estimated from the present data (Table 3, excluding San Juan). Thus, substituting $\sigma_1 = \sigma_2 = 10^{-6}$ and $|\mu_1 - \mu_2| = 10^{-7}$ into eq. (18), we find that $n_1 = n_2 = 200$ will give t = 1.99, which exceeds the value of t distribution, 1.97, for 198 degrees of freedom at 95% confidence level. Thus a variation in the range ratio of 10^{-7} found by averaging 200 ratio measurements is significant at 95% level of confidence.

At a rate of one measurement every hour, it will take slightly more than a week to complete this many measurements. Two such series of measurements one year apart is sufficient to determine the shear strain increment in southern California.

For a given set of t and o's, n's are approximately inversely proportional to the square of the difference in μ 's in equation (18). Thus doubling the measurement interval, thereby doubling the expected ratio variations, approximately quarters the required number of measurements. For example, a pair of 50-measurement sets two years apart will give the shear strain rate in southern California.

V. CONCLUSIONS AND RECOMMENDATIONS

Conclusions

Even though the field experiment we performed during this contract was quite limited compared with our original plan, we obtained several interesting and important results. The following is a list of conclusions drawn from these results:

- 1. The increment of the ratio of the length of a survey line to the average of several survey lines in a region is directly related to the incremental shear strain in the region. Thus, the shear strain rate can be calculated from observations of temporal variations in such ratios.
- 2. Using the TLRS, the time-of-flight ratios could be determined to an accuracy (one standard deviation) of 1×10^{-7} by averaging measurements over a four day period. This accuracy was obtained without using any atmospheric corrections at all. No improvement was obtained when atmospheric corrections based on end-point measurements were applied.
- 3. A calculation using a hypothetical data simulating the observed strain field in southern California indicates that two sets of TLRS ratio measurements separated by one to two years will be sufficient to determine the direction and rate of shear strain in the region.
- 4. Thus relative lateration using the TLRS has been demonstrated to be a good method for monitoring the regional shear strain field around satellite ranging stations. The TLRS operates successfully over long distances. The ratio method is extremely economical. It requires no environmental measureme: s and can be performed with small unattended retroreflectors distributed over a wide area. Thus these techniques greatly surpass the capability of conventional EDM techniques.

Recommendations

1. The results of the present experiment are thus very encouraging. However, they are based on only one experiment. Before this technique is put to a practical use, further demonstration is needed to confirm the above results. Therefore, it is recommended that this feasibility study be continued at least to include reoccupation of the Mt. Wilson site and two measurements at another properly selected site, preferably with a different meteorological environment.

- 2. Relative lateration is not limited to the data taken by the TLRS. The data reduction procedure used in the present study can be applied to other data from distance measurements. Therefore, it is recommended that we reanalyze some of existing ranging data to see if improvements in determination of shear strain rate can be achieved. This can be done without further field measurements.
- 3. Additional feasibility test measurements similar to the Mt. Wilson experiment may be obtained from fixed satellite ranging stations. It is therefore, recommended that this possibility be examined.
- 4. Horizontal ranging to distant targets on the ground does not require all the sophistication of the TLRS system. Therefore, when the capability of the present technique is fully demonstrated, a smaller, more portable single-photon ranging unit should be developed for this purpose.
- 5. Finally, the technology is advancing in other fields also. Such techniques as miniature interferometer terminals [Counselman and Shapiro, 1979] may someday be more useful in surveys of regional extent. Therefore, development in these other techniques should be reviewed while developing the present technique.

Acknowledgements

The initial data reduction of the Mt. Wilson experiment was done in the Astronomy Department of the University of Texas at Austin. We are grateful to Dr. Eric C. Silverberg for supplying us the processed data on a computer tape. Dr. Cliff Frohlich kindly reviewed a draft of this report, his constructive comments are greatly appreciated.

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APPENDIX

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Calibrated Round-Trip Time of Flight

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24 10 60 (30 5.9	693.1	6.0500	22 14 20 00	12.6	720.1	9.2050	22 20 32 00	19.2	្លាប់ ហ្គូបុ	2.3337
24 11 60 6	38 5.8 6.2	693.1 692.1	5.8176 5 8176	22 16 28 88 22 18 28 66	0 - 0	728.B	5、9131 10.6698	22 22 38 68	15.3		1.6974
	9 2 2 2 2 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	693.1	5.8176	22 22 90 00		719.3	11.8118	22 23 37 88	5.9 25	39.6	1.4747
24 14 88 6	38 5.5	693.1	5.5931	23 10 20 60	11.5	718.6		23 88 38 89	13.1 73	39.6 1	1.0553
24 15 00 6	J0 5.1	693.1	5.5931	23 16 20 00	12.1	720.8	9.5532	23 01 45 00	11.6 73	39.6 1	1.6553
24 16 00 (JØ 5.6	693.1	5.8176	23 18 20 00	12.4	720.8	10.2850	23 02 32 90	12.3 73	10.53 10.62	0.6688
24 17 00 (Jg 6.8	693.1	6.5400	23 28 38 88	14.7	720.8	11.8974	23 83 35 88	11.6		0.2850
24 13 00 (30 8.2	693.1	6.7978	24 88 39 88	11.3	720.1	9.3777	23 64 46 66		នេះ ភូម្ភ ភូម្	
24 19 69 6	30 9.3	693.1	7.2012	24 84 28 88 24 86 38 88		720.8	8.6/88 0 0670	83 85 CB 52 88 CZ 38 ZC		0 00 0 00	0.633U
			0C20.7	24 BC AF 68	יי סי		2 7361	23 87 38 86			9.9131
	11 DI		2300 8	00 Ct 00 t-7 00 Ct	ים יס	220.1	7.7791	23 88 38 88	14.1	1.02	9.5532
24 23 69 6		6-100 6-06-3	8.2263	24 10 10 00		720.1	B.2263	23 09 29 00	12.7 73	3.0 1	9.6688
25 00 00 6	10.3	690.8	7.3403	24 10 45 88	8.2	720.1	7.8610	23 10 35 00	12.4 73	8.5	0.2850
25 01 00 6	0 6.6	690.8	6.0500	24 12 10 00	9.8	720.1	7.7721	23 11 38 68	12.4 73	33.8	0.2850
25 02 00 6	JØ 5.8	690.1	6.0500	24 14 10 00	8.4	720.1	7.6259	23 12 40 09	12.5 73	20.03 20.03	8.2350
25 83 88 6	38 5.7	690.1	6.5400	24 14 45 60	8	720.1	7.6258	23 17 40 00	13.1 72	1.8	1.4747
25 04 00 6	30 5.3	690.8	5.8176	25 02 10 00	8. 5	719.3	8.5418	23 18 34 08	13.5 72	ω. 	3.2486
25 05 00 6	38 5.7	690.8	5.3762	25 02 52 00	ຜ ຫ	719.3	8.7036.	23 19 30 00	13.5 74		1.6653
25 06 00 6	18 5.7	690.1	5.1669	25 84 18 88	с. С	719.3	8.2263	23 20 40 60	16.3 72		2.5331
25 07 00 6	38 5.8	690.1	5.3762	25 86 10 98	0 4	720.1	6.7978	23 21 33 88	10.1 10.1 1		1.4/4/
25 88 89 6	38 5.6	689.3	5.3762	25 86 46 89	0 00 00	718.6	6.4143	23 22 30 88			1.4/4/
25 09 00 6	30 5.5	689.3	5.3762	25 88 10 60	œ (∧ (718.6	6.8500	23 23 26 00 24 29 20 20 00			1.8033 0171
25 10 60 6	38 5°.3	689.3	5.3762	25 18 18 68 27 81 92 75	່ ກໍເ	718.5 7	5./3/8 2 7070	24 60 36 60		- 0	
1 AB 11 42	15 2 2 2 2 2 2	089.5	0.0761 7770		0 0 7 0	710.5	0.1310 6 2908	24 82 48 68	10.7		9,2650
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2 14 60 5		689.3	5.3762	25 14 22 00	ی د ح	718.6	6.7978	24 84 35 88	10.7 74	11.8	9.2050
25 15 00 6	10 5.3	690.1	5.5931	25 16 10 00	11.4	719.3		24 05 40 00	18.3 74	8.11	9.2050
25 16 00 6	00 5.4	690.8	5.3762					24 05 20 00	10.3 74		8.8679
25 17 60 6	00 7.3	690.8	5.9327		Niguel		•	24 87 35 83	10.2 74	ະ ເຄື	8.8679
25 19 60 6	38 8.5	690.9	6.2908					24 03 35 00	5 2 5 7 5	8 · I	u. 5410
25 19 80 6	10.0	650.8	6.6678	22 09 28 00	13.2	738.8	8.8679	24 69 38 88	10.4 72	1.1	9. ZUCU
25 20 00 0	30 12.3	650.1	6.7978	22 10 32 60	12.5	738.8	9.2050	24 10 30 00	יי איו היי		6. J416
25 21 00 (30 16.2	689.3	8.2263	22 11 30 00	12.0	738.8	9.2050	24 11 39 00	5 5 7 7 7 7 7	1.1	6977-9
25 22 80 6	30 14.1	688.6	6.5480	22 12 30 00	11.4	738.8	9.2050	24 12 31 68		1.1	8. ZZ65
25 23 00 (16.1	689.3	7.3403	22 13 31 80	11.5	738.8	9.7318	24 13 33 88	יי ביו ביו	20	C. 3218
26 00 00 6	30 12.5	690.1	6.7978	22 14 30 60	14.1	739.6	9.9131	24 14 31 88	2 ~ r ~ r	ت ت م	7.6238
	1			22 15 41 68	13.0	740.3	9.5532	24 15 35 00		2 2	C. 3415 2555
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				22 17 30 40	16.2	740.0	11.8535	24 IC 20 00			1010.0
22 10 21 4	30 14.C	718.6	9.9131	22 18 51 VU	1.1	740.5	12.3335	24 18 1/ an		4 1 - 1	

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Time	н	۵.	^	Time	⊢	٩	>	Time	H,	۵.	>
dy hr mn sc	deg C	gHmm	mnHg	dy hr mn sc	deg C r	nnHg	mmHg	dy hr mn sc	deg C	6, eu	6Hinm
24 19 15 00	14.2	740.3	10.2650	22 14 49 98	9.8	23.3	5.3762	24 05 45 00	1.0	622.6	
24 28 23 88	16.0	739.9	11.0653	22 15 68 88	10.2 62	23.3		24 86 25 88	. 1.1	622.6	4.3764
24 21 40 60	16.1	738.1	10.6688	22 16 30 80	10.0 62	24.1	5.7043	24 86 49 68	1.2	622 . 6	4.2774
24 22 30 60	15.5	738.1	10.2850	22 16 45 00	18.8 62	24.1	5.5931	24 87 43 68	~ (82 1 -8	
24 23 30 00	15.4	738.1	10.2850	22 17 38 88	19.6	24.1		24 08 20 00	ю.	521-81 521-81	4.5311
25 88 27 88	13.1	738.1	9.0350	22 18 90 00	10.8 62	24.1	5.5400	24 09 20 03	ů.	621.8	4.4780
25 01 15 00	11.0	738.1	8.8679	22 18 45 60	13.0 62	24.1	7.3403	24 18 28 88	œ.	521.1	4.1783
25 02 27 00	9.4	738.1	8.2263	22 19 00 00	13.0 62	24.1		24 11 20 00	4.	621.1	
25 83 38 88	8.4	738.1	7.6258	22 20 00 00	12.5 62	23.3		24 12 20 00	0, 0,	521.1	3.8100
25 04 23 00	6.2	739.1	7.3403	22 21 60 60	13.0 62	22.6		24 13 20 60	ω. Ψ	521.1	
25 65 40 60	8.8	737.3	6.7978	22 22 60 60	11.6 62	22.6	5.2908	24 14 20 00	1.8	521.1	4.5811
25 86 25 88	8.9	737.3	6.7978	22 22 45 00	10.3 62	22.6		24 15 20 60	3.2	621.1	
25 87 28 88	ອ ເ	737.3	6.5400	22 23 88 88	11.0 62	22.6		24 16 28 88	2.2	521.1	4.0343
25 88 23 88	0. 7 0	736.6	6.7978	22 23 45 80	9.2	m. M	7.0645	24 17 28 88	м. М.	62 1. 8	
25 89 29 80	8.3	735.8	6.7978	23 00 00 00	9.2 62	21.8		24 18 20 00	3.4	521.8	5.0648
25 10 30 00	8.4	735.8	6.7978	23 00 45 00	7.2 63	21.8	5.0500	24 18 37 00	ы. В. В.	521.8	
25 11 35 80	8.5	735.8	7.0645	23 01 00 00	8.8 62	21.8		24 19 28 88	5.2 2.2	621.1	7.6253
25 12 29 88	8.Ø	735.8	7.3403	23 82 88 88	7.2 62	22.6		24 20 09 60	5.4	521.1	5.8176
25 13 30 00	6.2	735.8	7.0645	23 82 45 88	7.7 62	22.6 (5.5400	24 28 15 68	6.3 (521.1	
25 14 25 00	7.1	735.8	7.6259	23 83 68 68	7.3 62	22.6		24 20 30 00	7.16	520.3	
25 15 30 00	в . З	736.6	7.6258	23 84 98 88	6.2 62	22.6		24 20 40 60		528.3	6.4397
25 16 17 00	11.1	737.3	8.8679	23 84 45 88	6.2 62	22.6		24 28 58 88	7.5	520.3	
25 17 15 80	12.3	738.1	9.2050	23 85 88 88	6.5 62	22.6		24 21 00 00	00 00	520.3	
25 18 16 00	14.1	737.3	10.2850	23 06 00 00	6.5	22.6	5.4836	24 21 05 00		519.E	•
25 19 24 00	15.1	735.8	10.2650	23 86 49 83	6.3 62	22.6 (5.2988	24 21 15 88	2.9	619.6	
25 20 20 60	16.0	735.1	10.6633	23 07 00 60	7.0 62	22.6		24 22 60 69	8 	519.6	-
25 21 24 00	16.0	735.1	9.9131	23 08 00 00	7.3 62	21.8		24 22 15 00	0.0	519.6	7.8645
25 22 20 60	16.9	734.3	10.6698	23 08 40 00	7.7 62	21.1	4.9647	24 22 20 00	0. 0.	519.6	
25 23 26 00	15.3	734.3	10.2850	23 08 45 60	7.9 62	21.1		24 22 23 88	9.4 1	519.6	
26 89 28 89	14.4	735.1	9.9131	23 89 88 88	8.8 8.8	21.1		24 22 32 80	2°3	519.6	7.2012
				23 10 60 60	4.8 62	21.1		24 22 35 88	7.5 6	519.6	
	Sant lage			23 11 60 60	5.1 62	21.1		24 22 45 00	6.9	619 . 6	6.6937
		ļ		23 12 00 09	4.4 62	21.1	•	24 23 15 08	ਹ ਹ	519.6	
22 09 00 00	11.0	623.3		23 12 45 60	4.6	21.1		24 23 35 60		619.6 0	
22 10 90 90	10.3	623.3	5.5710	23 19 30 00	4°. 7°.	9.22	0.3762	00 CT 00 CZ	ָם קיי	0.010 0.010	0,000,0
22 10 40 00	10.1	623.3		23 20 00 00	5.1 62	22.6		25 88 28 88	4 I 9 (0.010 0.010	
22 10 46 00	10.2	623.3		23 21 00 00	2°2	21°8		25 88 25 89 25	ມ. ທີ່ ທີ່	014.0	
22 11 00 00	12.4	623.3		· 23 22 00 00	4.2			25 88 29 88	יי יי	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	
22 12 80 88	10.2	622.6		23 23 80 88	2.2	8. 		25 81 15 88			
22 12 42 66	10.5	623.3		24 88 66 88	1.4 62	21.8		25 01 20 00		2 C 2 C	1 2000
22 12 46 00	10.3	623.3		24 81 45 88	1.8 62	8.1.9		20 21 20 22 20 20 20 20	4 P V C		4.1030
22 12 50 00 22 12 50 00	10.3	623.3		24 82 45 88 24 52 45 66	3 C 3 C	ה. היו		מט וז זט נז שט כב כס שכ	າບ ບໍ່ດ	0 10 10 10	
22 13 60 00 33 14 00 00	9.0 9.0	623.3		00 27 70 72 00 27 00 27 00 20 00 20 00 00 00 00 00 00 00 00 00	20 0 0 1	יים גיים גיים	A Z76A	מט אב אט בא	2 M C		
22 14 20 20	מ ת	072.0		24 84 43 55	0.		10.0.1	לט מני מה כני	·	2.0	

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Table A2. (continued)

Timo	F	<u>م</u>	>	Time	F	۵.	>	Time	⊢	۵.,	>
dy hr mn sc	deg (; mmHg	mmHg	dy hr mn sc	deg C	pHnim	maHg	dy hr mn sc	deg C	maHg	gHam
25 82 43 88	2.3	618.8	•	26 89 29 08	រ រ រ	617.3	6.1694	24 04 45 09	ы С	719.3	8.3828
25 02 51 90	2.4	619.8	4.7315	•				24 65 09 30	ທ ຜ	719.3	8.2263
25 03 00 00	2 2 2	613.8	•		San Jua	c		24 85 38 88	ເດັດ ເດັດ	719.3	8.8724
25 83 15 68	1.7	613.9				, i		24 96 37 98	ອ່ເ ໝໍດ		1 0010
25 63 27 80	1.5	618.8		22 88 80 88	16.0	717.8	9.0048	24 85 25 88 24 15 55 66			010010 1 12000
25 04 15 00		618.8	4.5811	22 09 03 00	15.8	717.8	8.5418 5.4255		ລຸດ ກັດ		
25 04 25 00	. .0	618.8		22 10 02 00	15.3	717.8	9.1366	24 10 10 14 23	ລູດ ວັດ		
25 85 15 88	ب	618.8	•	22 10 58 80	14.8	217.8	11.0251	24 12 88 68	ສຸເ ໝູ		,
25 05 25 00	æ.	618.8		22 11 58 89	16.0	717.8	9.6957	24 13 18 68		718.0	
25 86 15 80	1.8	618.8		22 12 59 08	16.5	717.8	10.6299	24 14 68 69		717.8	7.8535 1
25 86 23 88	2.1	618.8	5.5811	22 14 26 00	15.8	717.8	11.0653	24 15 82 88	2.0	212-8	7.0645
25 86 26 88	2.0	613.8		22 15 83 88	16.0	718.6	7.8619	24 15 03 08	ເລຍ ເລຍ ເລຍ	718.6	
25 86 35 88	1.9	618.8		22 16 05 00	ររ រ រ	719.3	7.8016	24 16 33 88	ກເ ກ		1 10110
25 86 37 88	1.9	618.8		22 17 62 69	15.8	219.3	B.1 643	24 17 31 88			
25 86 50 80	2.1	618.8		22 18 80 60	19.3	719.3	8.5418	24 18 29 88	N (21.7.0 21.1	
25 07 15 00	5 1 2	618.8		22 19 86 60	25.6	719.3	10.2850	24 19 34 88	ור. פיי		5.8402
25 07 25 00	2.8	618.1		22 20 02 00	22.2	718.6	10.2850	24 20 33 59	16.5	10.5	9.73.0
25 08 15 00	9	618.1	4.3764	22 21 00 00	20.0	717.1	10.6688	24 21 30 06	10.01	716.3	9.6358
25 08 20 00	2	618.1		22 22 80 80	18.5	717.1	11.4747	24 22 30 86	10.0	715.0	9.5332
25 88 24 68		618.1		22 23 60 69	13.0	717.8	11.3916	24 23 30 00	14.5	715.6	9.0350
25 89 26 86	1.1	617.3		23 88 88 88	15.0	717.8	10.0975	25 09 30 09	12.0	715.6	8.0724
25 10 15 00	1.9	617.3		23 81 88 89	13.5	717.1	10.4752	25 01 30 00	 ເກ	715.6	8.8724
25 18 28 68	1.8	617.3		23 82 88 88	12.0	717.8	9.9131	25 82 36 88	ເ ເ ເ ເ ເ ເ ເ ເ ເ ເ ເ ເ ເ ເ ເ ເ ເ เ	715.6	7.9218
25 18 23 88	1.8	617.3	4.2774	23 83 88 88	11.0	718.6	9.9131	25 83 38 88	0 0	715.6	7.6258
25 18 35 88	1.6	617.3	, , ,	23 84 10 68	12.0	718.6	9.5532	25 04 35 89	ຕີ ເ	715.6	7.3618
25 IN 38 NO		617.3		23 85 89 88	11.5	718.6	9.7318	25 05 30 60	ເ ດີ ເ	715.6	c. 2383
25 18 58 88	1.8	617.3		23 86 80 69	11.0	717.8	9.5532	25 86 35 89		715.6	7.66.15
25 11 15 80	2.1	617.3		23 87 88 60	11.5	717.8	9.5532	25 87 38 88	2.2 2	714.8	7.2012
25 11 20 00	2.1	617.3		23 08 02 00	12.0	717.1	9.3881	25 88 32 89	ខ	714.1	6.0678
25 12 15 00	6	616.6	3.8100	23 69 69 60	12.0	716.3	9.0350	25 69 30 66	6 6	713.3	6.3894
25 12 20 00	^m	616.6	4.3764	23 10 00 00	10.5	717.1	9.2392	25 18 31 89	10.0	713.3	6.1935
25 12 23 00	4	616.6		23 10 56 60	11.0	717.8	9.5179	25 11 30 88		715.5	6.60/8
25 13 15 00	9.	617.3		23 12 83 88	11.5	717.8	9.4475	25 12 31 89		210.0	6.7474 1.774
25 13 20 00	ມ	617.3		23 18 80 60	12.5	719.3	9.1026	25 13 35 88		-14° F	1. 2000 1
25 14 20 00	ю	617.3	4.3764	23 19 88 88	15.5	719.3	9.5532	25 14 29 08		714.1	6.9428 1
25 15 20 00	1.9	618.1		23 20 85 88	15.5	718.6	10.2850	25 15 32 88	ы. 1	74.8	
25 16 20 00	4.7	618.1		23 21 00 60	13.0	718.6	9.9131	25 16 32 69	11.0	715.6	8.5553 0.7553
25 17 20 00	5.5	618.8		23 22 10 60	14.5	717.8	9.7318	25 17 33 80	17.8	715.6	9.2026
25 18 20 00	6.2	618.8	6.0500	23 23 05 08	12.0	717.8	9.2858	25 18 33 80	4 4 1	3.412	8.4416 9.4416
25 19 20 00	7.7	618.1		24 00 00 00	11.5	718.6	9.2050	25 19 30 00	רי הית וויית	(14°1	00000
25 20 20 00	8.3	617.3	6.2908	24 01 00 00	11.5	718.6	9.0350	25 20 35 88	18.5	5.0 10 10	d. Jaza J
25 21 20 00	9.4	617.3		24 02 00 00	6 0	719.3	0.6305	25 21 45 09	10 20 20 20		
25 22 20 00	8.8	617.3		24 83 80 68	9 9	719.3	8.2263	25 22 33 UU	ទា		
25 23 20 00	6.8	617.3		24 84 68 58	9.0	719.3	8.0724	בכ טט כב	10.0	0.01	

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Table A2. (continued)

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Time dy hr mn sc	E	Time dy hr mn sc	c	Time dy hr nn sc	c	Time dy hr mn sc	c
M W		24 13 54 08 24 14 28 00	1.00025008 1.00025008	26 01 09 09 27 21 13 23	1.00024528 1.00024919	24 89 89 88 24 19 69 68 24 19 69 68	1.63027267 1.03327286
22 12 08 01	1.00024386	24.14.54.16 24.15.53.54 24.15.43	1.86025848 1.86025848	Cast	0	24 11 48 88 24 12 89 88 24 13 68 88	1.88327276 1.88327276
22 13 08 54 22 14 52 40	1.80024473 1.60024414	24 15 41 49 24 17 53 25	1.000242033	 27 89 18 88	1.00026392	24 13 00 00 24 14 89 68	1.03027325
22 15 59 57	1.80024414 1.80024414	24 If J3 29 24 19 06 29	1.00024679	22 10 00 00	1.00026383	24 15 09 09	1.00027365
22 16 31 16	1.00024327	24 19 55 54	1.00024679	22 11 00 60	1.00025443	24 15 83 98	1.68827315
22 16 57 27	1.00024414	24 20 49 38	1.00024558	22 12 00 00 22 12 00 00	1.60026465	24 17 08 69 24 10 66 66	1.05027198 1.000020198
22 17 59 44 22 18 14 19	1.00024272 1.00024071	24 21 24 25 24 27 22 22	1.00024411 1 85824411	22 13 00 00 22 14 00 00	1.00026493	24 10 30 68	1.68326958 C
22 21 57 23	1.00024183	24 22 52 51	1.86924411	22 15 00 00	1.00025502	24 28 83 89	1.00026747
22 22 25 11	1.00024129	24 23 55 10	1.00024588	22 16 00 00	1.00026447	24 21 09 66	1. 10326545
22 23 02 15	1.02024183	25 00 34 16	1.00024499	22 17 89 89 22 10 58 68	1.00026459 1.00026256	24 22 33 69 24 27 00 69	
23 88 88 56 23 88 28 25	1.80824269 1 88824255	22 22 24 24 25 21 29 25 25	1.68824766 1 88824857	00 00 01 22	1.00020370 1.00026294	25 88 88 88 88	1.83825773
23 00 20 20	1.00024355	25 83 55 51	1.06624948	22 20 00 60	1.00025994	25 01 09 69	1.00027129
23 02 01 37	1.00024442	25 84 32 31	1.00024857	22 21 00 00	1.00026228	25 82 83 88	1.00027178
23 82 24 65	1.00024442	25 05 54 44	1.00024857	22 22 80 80	1.00026328	25 83 88 80	1.03827187
23 02 48 26	1.00024529	25 86 28 44	1.00024857	22 23 88 88	1.66026237	25 84 69 68 25 84 69 68	1.00027215
23 84 13 88	1.00024529	20 10 70 22 25 23 26 30	1.02024637 101020000	23 00 00 00 27 01 00 00	1.00026411 1 00026522	25 R6 R9 R9	1.66727187
23 84 33 54 23 85 57 22	1.00024619	27 00 70 42 25 08 36 19	1.66624887	23 82 88 68	1.00026632	25 07 00 09	1.62027256
23 06 37 07	1.00024618	25 09 48 16	1.05024796	23 03 00 00	1.06026764	25 68 89 83	1.68327166
23 06 55 29	1.00024529	25 10 28 22	1.00024706	23 84 80 68	1.00026615	25 89 83 88	1.00627175
23 07 52 58	1.00024550	25 11 60 51	1.00024678	23 85 88 88	1.00026634	25 10 90 93 25 11 80 69	1.96927195
23 88 28 32	1.00024590 	25 11 59 30 25 12 24 14	1.00024859 1.00024859	23 86 88 88 23 16 88 89	1.08026643 1.06077185	00 00 11 22 00 00 12 22	1.02727254
23 89 56 13	1.80024679	25 13 57 68	1.00024678	23 17 00 00	1.00027034	25 13 00 00	1.08927148
23 10 29 17	1.00024679	25 14 26 09	1.00024768	23 18 00 00	1.00026863	25 14 63 60	1.00327265
23 10 55 37	1.00024736	25 14 56 49	1.00024796	23 19 88 88	1.00026684	25 15 83 89	1.63927227
24 04 06 03	1.00025040	25 15 55 10	1.69824738	23 20 60 00	1.00026882	25 16 88 50 25 17 50 06	I. BUGZIZGA
24 04 22 39	1.00025132	25 16 36 22 25 13 55 22	1.00624649 • 80624471	23 21 00 00 27 22 60 60 60	1.0002555C	22 10 00 51 CZ	1.05026945
27 CC 40 47	1.00023040		1 96674364	23 23 88 88	1.00026803	25 19 66 60	1.05526802
24 BG 34 27	1.00025132	25 19 85 52	1.00024352	24 88 88 88	1.00026965	25 28 68 88	1.60326559
24 07 04 10	1.00025040	25 19 55 15	1.00024265	24 01 00 00	1.00027119	25 21 68 83	1.00025171
24 07 51 41	1.00025132	25 20 31 54	1.00024265	24 02 00 00	1.00027255	25 22 88 69	1.03026335
24 88 31 26	1.00025132	25 21 06 50	1.00024179	24 03 00 00	1.00027264	25 23 83 69 25 23 83 69	1.42325188
24 89 46 28	1.60025040	25 21 54 44	1.00024238 . 00024153	24 84 88 88 24 85 88 88	1.88827296 1 88827285	26 88 88 89	1.8885558
24 10 28 48 24 11 00 43	1.00025132	25 23 82 87	1.00024238	24 86 88 88	1.00027276	San Pe	dro
24 11 55 50	1.00025132	25 23 48 19	1.00024528	24 07 00 00	1.00027275		
24 12 34 12	1.00025100	26 00 36 21	1.03024352	24 88 89 88	1.00027255	22 10 21 89	1.00927425

Table A3. Group Index of Refraction at End Points

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	Table A3.	(continued)				*****	
Time	c	Time	c	Time	c	Time	c
dy hr mn sc		dy hr nn sc		dy hr nn sc		dy hr mn sc	*******
22 12 20 00	1.00027576	22 19 33 88	1.00027924	24 19 15 88	1.00028302	22 14 49 68	1.05024260
22 14 20 60	1.80027684	22 20 32 00	1.00027792	24 20 23 00	1.00028369	22 15 00 69	1.00024166
22 16 20 00	1.00027397	22 ·21 40 88	1.00028002	24 21 40 60	1.88826933 1 60030531		1.05027214
22 18 20 00	1.80027095 1.00027095	22 22 26 66 27 25 26 66	1.000202055	00 02 27 77 70	1.00028101	22 12 20 00 22 17 20 00	1.66324214
00 00 27 77	1.00027777 1.000377777	00 35 00 25	1 00028294	25 88 22 88	1. PER28327	22 18 00 00	1.63024146
22 16 20 60	1 00027260	23 80 35 88	1.66628534	25 01 15 00	1.00028536	22 18 45 60	1.66823968
23,18,28,68	1.00627731	23 82 32 88	1.80828464	25 02 27 00	1.02028698	22 19 68 69	1.63023960
23 20 30 00	1.00027509	23 03 35 00	1.69928534	25 83 38 88	1.00028800	22 20 03 69	1.65823971
24 00 20 00	1.86827811	23 04 40 00	1.00028524	25 04 23 00	1.00028851	22 21 69 63	1.65323983
24 84 28 88	1.00028055	23 85 38 88	1.80028534	25 05 40 00	1.00028728	22 22 90 98	1.63924820
24 06 10 00	1.00026925	23 06 32 00	1.00028543	25 86 25 88	1.00028717	22 22 45 80	1.68024158
24 06 45 00	1.00028035	23 07 30 00	1.00028533	25 07 28 00	1.00028758	22 23 89 80 20 21 11 20	1.83324371
24 68 10 68	1.00028038	23 08 30 00	1.00928223	25 08 23 09	1.00028711	22 23 45 HB	1.66424291
24 10 10 60	1.00028107	23 69 29 88	1.00328393	25 09 29 60	1.00028720	25 89 88 88	1.60024145
24 10 45 00	1.00028117	23 10 35 00	1.00028423		1.00028710	23 UE 40 US	
24 12 18 60	1.00028977	23 11 30 00	1.60028423	25 11 35 88	1.00028760	25 81 88 83	
24 14 18 88	1.00028097	23 12 49 89	1.00028413	25 12 29 00	1.000000.1	23 67 68 68 24 02 78 68	1.0222000
24 14 45 80	1.00028127	23 17 40 60	1.86028469	25 13 38 88	1.00026761		
25 82 18 88	1.00027927	23 18 34 88	1.09628429	25 14 25 UU	1. UUUZUS45		
25 02 52 80	1.00027927	23 15 38 88	1.63623429	25 15 30 60	1.00028698	23 64 69 69	
25 04 10 00	1.00027927	23 20 49 00	1.60623128	25 16 17 88	1.000284995	23 84 45 68	
25 86 10 00	1.00027998	23 21 33 88	1.00028157	25 17 15 00	1.80828496	25 82 64 88 27 27 28 69	SCORESSIN .
25 06 46 00	1.00028019	23 22 30 60	1.500282893	22 18 18 CZ			
25 88 18 88	1.00028099	23 23 26 88	1.69928137	25 19 24 60 25 26 26 26	I. 83828345 • 88637836	23 02 20 23 33 37 30 60	
	1.00027999	24 88 38 88	1.000000		1.8002/323 1.60037030	22 60 87 89	1 909003557
25 10 45 60	1.00027999	24 61 46 60	1.00230710 1.00030710	50 57 17 CZ	1.00027212	23 82 24 89	1.60624295
	1.88825833 1.000007	24 82 43 88 24 82 58 88	1.UUUKAK10 1 8087869	22 22 20 00 22 22 C2	1.80027955	23 82 45 80	1.05824273
	1.0002001	24 60 20 00 00 00 00 00 00 00 00 00 00 00 00	1 00000210	26 68 28 88	1.00025984	23 69 86 60	1.63324269
22 14 22 00	1.00027770	24 84 33 88 24 85 48 80	1 06028750			23 10 99 50	1.0822-3545
		24 86 29 89	1.80828758	Santi		23 11 69 60	1.00024522
Nia	uel	24 07 35 00	1.00023760			23 12 68 89	1.03924534
	 	24 08 35 00	1.99028311	22 09 00 00	1.06024098	23 12 45 58	1.959245565
22 89 28 80	1.00028344	24 09 30 00	1.93828713	22 13 88 88	1.00024157	23 19 30 69	1.00024599
22 10 32 00	1.60028413	24 10 30 00	1.90028835	22 10 49 40	1.60824174	23 28 83 83	1.0042059.1
22 11 30 60	1.00028463	24 11 39 00	1.80028906	22 18 46 88	1.00024166	23 21 EU 85	1.00024514
22 12 30 00	1.00028523	24 12 31 00	1.00028917	22 11 60 00	1.00023980	23 22 06 68	1.00024550
22 13 31 00	1.00028513	24 13 33 88	1.00026927	22 12 88 88	1.60024139	23 23 99 68	1.58624888
22 14 30 00	1.68028286	24 14 31 80	1.00028958	22 12 42 80	1.08024140	24 03 33 03	1.63924300
22 15 41 80	1.00028362	24 15 35 00	1.00028793	22 12 46 60	1.00024157	24 81 45 88	1.13024917
22 16 36 00	1.69928224	24 16 17 00	1.06028693	22 12 50 0U	1.00024157	24 82 45 80	1.2302401C
22 17 30 00	1.00328107	24 17 20 00	1.00026392	22 13 68 88	1.00024183	24 65 40 80	Ususacas.
22 18 31 60	1.88828920	24 18 17 80	1.80828432	22 14 80 66	1.00024269	24 64 45 AU	I.TJUZADOC

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	Table A3. (cont inved)			0 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8		
Time	2	Time	c	Time	c	Time	c
dy hr mn sc		dy hr mn sc		dy hr masc		dy hr m sc	
24 85 45 88	1.00024949	25 02 43 00	1.38324679	26 00 20 00	1.00024337	24 84 45 68	1.63928956
24 06 25 09	1.80024940	25 02 51 00	1,00024570			24 85 89 69	1.63029656
24 86 48 88	1.00024931	25.03 00 00 25.23 15 00	1.39024679	San	nan	24 65 38 88 24 65 77 69	1.00020165
24 67 48 66	1.80024944	20 CI 50 CZ	1.00024735 1 36672751	22 83 88 88	1_8682221	24 68 28 83	1.63525945
00 07 00 77 00 07	1.00024333	25 84 15 88	1.30024787	22 89 83 88	1,00027290	24 10 85 88	1.20327979
24 10 20 00	1.00824907	25 84 25 88	1.50024796	22 10 02 00	1.00027338	24 11 81 88	1.66626379
24 11 20 00	1.60024943	25 05 15 80	1.30024842	22 10 58 80	1.00027385	24 12 53 60	1.60225979
24 12 20 00	1,00024907	25 05 25 00	1.06024815	22 11 59 80	1.60027271	24 13 18 60	1.86228379
24 13 20 00	1.00024907	25 86 15 88	1.00024724	22 12 59 88	1.00027224	24 14 88 86 24 15 22 68	1. ESUZUCUT
24 14 20 60	1.00024816	25 06 23 00 25 67 27 68	1.UU324697	22 14 26 00 22 15 02 00	1.00027202 1 00027202	00 70 CI 77	1.00026459
00 07 CI 77	1.80824690 1 86624798	00 07 00 C7 00 52 90 56	1.00024705	22 1J 03 00 22 16 85 88	1.00027376	24 16 33 68	1.00025319
24 17 20 00	1.00024692	25 86 37 88	1.66024715	22 17 82 88	1.00027347	24 17 31 66	1.68827733
24 18 20 00	1.00024700	25 86 50 88	1.00024697	22 18 60 60	1.00027020	24 18 29 60	1.66827538
24 18 37 00	1.00024665	25 07 15 09	1.00024679	22 19 00 00	1.00026450	24 19 34 63	1.03027339
24 19 20 00	1.00024513	25 07 25 00	1.00024607	22 20 02 00	1.00026729	24 20 30 60	1.03327168
24 20 00 00	1.00024495	25 88° 15 88	1.00024805	22 21 68 68	1.00026873	24 21 36 68	1.68027262
24 20 15 00	1.83024417	25 08 20 00	1.66624841	22 22 00 00	1.00027011	24 22 58 36	1.50227128
24 20 30 00	1.00024316	25 88 24 88 27 28 29 88	1.00024850	22 23 69 69 27 29 69 69 69	1.88827854 • 88877555	24 23 30 00 35 08 70 30	
24 20 40 90	1.60024298	22 29 22 29 22	1.00024/2/	27 01 00 00 00	1.00072000	25 01 23 50	COULTERON 1
24 28 58 68 24 21 68 68	1.00024281	00 CI 01 CZ	1.00024633	27 07 68 68 68	1.00021402 1 89822654	25 82 35 88	1.66827912
24 21 05 00 24 21 05 00	1.00024753	25 18 23 88	1.00023004	23 83 88 88	1.00027782	25 63 39 09	1.03327961
24 21 15 80	1.86824219	25 18 35 88	1.00024692	23 04 10 00	1.60927685	25 04 35 60	1.60027912
24 22 02 08	1.00024150	25 18 38 89	1.00024673	23 05 00 60	1.00027733	25 85 39 68	1.65527813
24 22 15 00	1.00024133	25 10 50 00	1.60024664	23 06 00 00	1.00027751	25 66 35 68	1.65327942
24 22 20 00	1.60024142	25 11 15 60	1.60024638	23 87 88 88	1.00027703	25 87 38 88	1.0000000
24 22 23 00	1.68024898	25 11 20 80	1.89824638	23 88 62 88 23 80 60 60	1.80027627 . 80037507	22 89 22 80 25	1.166037772
24 22 32 88	1.06024228	25 12 15 60 75 12 15 60	1.00024/20 1.00024722	00 00 60 57 22 10 00 00 22	1.00027275 27777000	25 10 31 60 25 10 31 60	1.66827675
24 22 25 00 24 22 25 69	1.00024235	25 12 23 88	1.88024762	23 10 56 00	1.00027751	25 11 30 80	1.63327921
24 23 15 88	1.00024436	25 13 15 00	1.00024772	23 12 03 00	1.00027703	25 12 31 69	1.65527921
24 23 35 60	1.00024471	25 13 26 68	1.60024782	23 18 83 88	1.00027663	25 13 35 00	1.69928592
25 00 15 00	1.00024502	25 14 20 00	1.00024854	23 19 60 60	1.00027376	25 14 25 68	1.86028302
25 88 28 68	1.00024560	25 15 20 00	1.00024687	23 28 85 88	1.00027349	25 15 32 68	1.66827882
25 60 25 60	1.60024578	25 16 20 00	1.88824439	23 21 88 88	1.60627588	25 16 32 60 27 15 37 60	1.83327666
25 68 29 68	1.00024537	25 17 20 00	1.60024396	23 22 10 00	1.00027414		1.05027054
25 01 15 00	1.00024635	25 18 20 00 27 10 20 20 20	1.00024555	00 C0 57 57	1.002/2000.1	00 02 01 CZ	1 0307178
25 01 23 00 25 23 5 60	1.00024644	22 19 28 28 29 25 26 36	1.00024178 • 00024025	24 00 00 00 24 01 00 00	1.00027723	25 28 35 88	1.88826914
00 CT 20 CZ	1.00024050 1.00027670	22 21 28 80	1 000000001	24 02 00 00	1.00027957	25 21 45 89	1.00325842
25 82 32 88	1.00023053	25 22 28 RM	1.66624652	24 03 00 00	1.00028006	25 22 33 60	1.03626764
25 02 39 00	1.00024679	25 23 20 00	1.00024224	24 84 88 88	1.00028006	26 60 35 00	1.08327654

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		Tab	le A4. Ti	me-of-F1ight	Rat ios				
Castri		San Ped	0	Nigue1		Sant iag	0	San Juar	_
dy hr mn sc	1,104+	dy hr mn sc	0 . 948+ ·	dy hrmnsc	1.364+	dy hrmn sc	1.210+	dy hr m sc	0.735+
22 18 84 29	914198	22 18 18 55	870439	24 04 20 32	486752	22 10 41 41	567622	22 18 51 17	646886
22 12 02 41	915120	22 12 21 52	871345	24 06 19 58	487793	22 12 43 11	567332	22 12 51 51	645954
22 14 05 43	914546	22 14 15 03	870661	24 86 46 28	485566	22 14 34 32	566961	22 14 48 84	647619
23 88 89 48	913745	22 18 23 39	871637	24 08 18 26	486255	22 22 49 55	566262	22 18 55 83	643189
23 02 09 25	913337	24 84 28 48	870474	24 10 15 13	486647	23 00 41 06	566832	22 22 54 50	5-8337
23 06 05 03	913747	24 06 16 19	871245	24 10 40 16	436857	23 84 33 84	566222	23 05 47 81	64, 31,5
24 06 03 46	914336	24 86 51 84	876690	24 14 15 10	486632	23 85 39 34	565878	23 68 47 14	647235
24 06 58 17	914260	24 08 69 40	871172	24 14 40 18	486914	23 88 39 44	565151	24 84 45 14	643393
24 08 60 19	913737	24 10 10 03	870674	24 16 18 20	486108	24 04 40 08	566802	24 85 29 45	647393
24 89 59 54	914011	24 18 45 34	870703	25 82 21 58	486409	24 86 25 43	566683	24 86 35 37	649839
24 10 55 03	914255	24 12 19 84	870911	25 82 43 49	485806	24 06 41 31	565959	24 BB 22 BB	645657
24 12 00 58	914457	24 14 11 30	870931	25 10 16 05	486836	24 08 22 17	566526	24 12 25 88	64304C
24 14 01 34	914142	24 14 45 60	870330	25 10 41 13	486512	24 18 21 13	566338	24 14 25 28	640772
24 14 55 05	915268	25 02 13 18	870356			24 18 35 15	566618	24 14 38 64	649358
24 16 01 03	915165	25 02 53 27	871173			24 12 20 03	566316	24 16 20 54	649338
24 18 00 06	914538	25 04 09 53	369977			24 14 20 03	564893	25 82 33 83	643673
24 18 57 39	914327	25 86 19 85	870052			24 14 35 11	564750	25 82 36 55	0420034
24 22 44 42	913891	25 06 45 43	870209			24 16 29 53	565410	25 84 25 68	653336
25 82 85 25	914683	25 08 11 36	871274			24 23 18 84	565626	25 85 32 42	642312
25 02 59 84	914826	25 10 10 16	871390			25 00 25 13	565889	25 83 23 56	647659
25 64 60 52	913718	25 10 45 23	870534			25 82 29 38	565803	25 13 25 11	647534
25 86 81 17	914789	25 12 10 30	871738			25 02 39 51	565623	25 10 39 93	647.275
25 86 55 21	915255	25 14 15 11	871417			25 84 22 37	566544	25 12 25 03	6482333
25 68 64 49	914402	25 14 48 42	871316			25 06 22 51	566702	25 18 31 55	647344
25 10 01 53	914614					25 86 35 48	566195	25 20 25 59	643656
25 10 55 08	915208					25 03 20 32	566493	25 22 28 53	643712
25 12 04 05	914277					25 10 20 08	566462	26 60 32 50	640333
25 14 51 54	915409					25 18 35 11	566862		
25 16 00 28	914909	~				25 12 21 02	565658		
25 18 80 19	914473					25 14 21 41	566251		
25 18 59 43	914613					25 14 35 02	566204		
25 20 07 42	914243	• ~				25 16 20 06	567037		
25 28 37 24	913952					25 18 39 42	566858		
25 20 49 58	914766	·				25 28 28 59	556413		
25 22 51 57	912786		٠			25 22 37 54	565985		
26 00 05 27	912961					26 00 26 53	565337		

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Fig. A2. Observed atmospheric pressure.

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