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An RC-1 Organic Rankine Bottoming Cycle for an Adiabatic Diesel Engine

L. R. DiNanno, F. A. DiBella, and M. D. Koplow Thermo Electron Corporation

December 1983

Prepared for NATIONAL AERONAUTICS AND SPACE ADMINISTRATION Lewis Research Center Cleveland, Ohio Under Contract DEN 3-302

for U.S. DEPARTMENT OF ENERGY Conservation and Renewable Energy Office of Vehicle and Engine R&D

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L. R. DiNanno, F. A. DiBella, and M. D. Koplow Thermo Electron Corporation Waltham, Massachusetts 02254

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Dedication to Grover C. Kinsman

Those of us who worked on Bottoming Cycle Systems at Thermo Electron Corporation have long recognized Grover Kinsman as one of the leading designers, for his contributions were many and significant. More than a valued and competent employee, he was a person who could be counted upon for help when needed. His reasoned judgment and willing manner made working with him both pleasant and productive. It is difficult to do justice to the magnitude of his contribution.

We are all saddened and humbled by his untimely death and he will be greatly missed. The memory of our good friend - his warmth and humanity - will be with us forever.

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SUMMARY

The major effort and work accomplished in this contract with NASA Lewis Research Center, as part of their Waste-Heat Utilization Programs, are highlighted below.

- A total of 1627 hours of RC-1 testing was completed in the dynamic loop at the following operating temperatures:
 - 442 hours at 700°F
 - 653 hours at 800°F
 - 532 hours at 900°F

The methods of sample analysis included:

- Neutralization Number by Color-Indicator Titration (performed in-house)
- Gas Chromatography (performed in-house)
- Mass Spectral Analysis (performed by an outside laboratory)

Static capsule tests up to 900°F temperatures were also performed in a parallel effort. The results of all tests performed, both dynamic and capsule, showed no thermal degradation or changes in RC-1 fluid constituents.

• A system analysis was performed. Data on adiabatic diesels in configurations both with and without aftercooling were supplied by NASA Lewis, and the performance of an RC-1 Rankine bottoming cycle system was computed for these engines.

The analysis showed that the bottoming cycle effectively utilizes the relatively higher exhaust energy of non-aftercooled diesels to the extent that increased bottoming cycle power output more than compensates for a slight diesel performance penalty associated with non-aftercooling. Bottoming cycle power output reaches 56 horsepower, which represents a 17.6-percent increase in the 317 horsepower of a turbocharged non-aftercooled diesel engine (TC) to yield a compound power output of 373 horsepower. The calculated brake specific fuel consumption (BSFC) for this diesel with a bottoming cycle system is 0.268 lb/bhp-hr, or 8.5percent better than the comparable turbocompound-aftercooled engine used as the diesel performance reference. Application of the bottoming cycle to a turbocompound engine produced a BSFC result of 0.258 lb/hp-hr, or 12-percent better than the turbocompound-aftercooled diesel.

- A system design for a typical truck installation was prepared and incorporates all components of the RC-1 Rankine-cycle system into three (3) subsystems: the power conversion unit (PCU), which includes the turbine, gearbox, and feedpump; the vapor-generator module; and the condenser-regenerator module. The condenser-regenerator module consists of a regenerator that has been integrated with the cylindrical aircooled condenser. The cooling module also contains an electric motor with a double-ended through-shaft that drives the booster pump, gerotor lubrication pump, and the clutched fan.
- A life-cycle cost analysis was performed for the RC-1 system design and the potential capital cost of the unit was estimated at \$8400 (selling price). Using the NASA reference data provided (i.e., selling price/manufacturing price = 2.0, fuel price = \$1.20/gallon), the simple payback period of a turbocharged diesel plus bottoming cycle (without taxes or maintenance) was calculated to be just under 3 years, when compared to the reference turbocompound-aftercooled diesel. The payback interval decreases as fuel price increases.
- Areas of technical development that have been defined as a result of this program, and need to be addressed are:
 - Heat exchanger fouling

- RC-1 working fluid thermal stability at elevated temperatures up to and exceeding 1000°F
- High-temperature seal development.

1. INTRODUCTION

The steep increases in the price of transportation fuels over the last decade have spurred research and development of more efficient prime movers for heavy-duty transport equipment. A major thrust of this work has been the improvement of the direct injection diesel engine the most efficient and ubiquitous power source for heavy-duty mobile applications. Incremental improvements in engine efficiency have been brought about through refinements of conventional diesel engine technology to improve thermodynamic performance and combustion phenomena and reduce engine friction. A more innovative approach to improved diesel engine efficiency - one that offers a substantial improvement in efficiency - is the adiabatic engine concept. Through the use of hightemperature materials (mainly ceramic compounds with exceptional mechanical properties at elevated temperatures) in those parts of the engine exposed to the combustion process, heat loss from the engine is greatly diminished and there is a resultant increase in engine efficiency. An increase in exhaust gas temperature is characteristic of the adiabatic engine. Only a part of the extra heat energy contained by the adiabatic engine can be converted to work. The balance of the heat is carried from the engine in the higher temperature exhaust gas.

The other approach to improving prime mover efficiency is the concept of engine compounding wherein a second prime mover is employed to recover power from the reject heat of the fuel-consuming prime mover in this case the exhaust gas of the diesel engine. Over the past decade, Thermo Electron Corporation has been at the forefront of Bottoming Cycle Technology. We have developed and tested a fully operational diesel/ organic Rankine-cycle compound engine for heavy-duty transport that has shown fuel savings of more than 14 percent over the baseline diesel engine alone in dynamometer tests, and over 13 percent in on-highway vehicle tests (ref. 1). These results have been obtained by bottoming present heavy-duty diesel engines that have exhaust temperatures in the range of 650° to 900°F (depending upon their size, fuel air ratio, degree of turbocharging, and whether they are two cycle or four cycle). For this temperature, Fluorinol, a mixture of trifluoroethanol and water, has been shown to produce the greatest power recovery of any working fluid. However, because the practical upper temperature limit for Fluorinol is about 600°F (with conditioning), due to thermal decomposition, it is not the optimum organic working fluid for the higher temperature exhaust gas of the adiabatic diesel engine. These higher temperature heat sources call for a working fluid with higher temperature capability.

In this waste-heat utilization program to design a Rankine-cycle system to bottom an adiabatic diesel engine, the organic fluid designated RC-1 was chosen as the working fluid. The program included tasks to conduct a system analysis, preliminary design, and cost analysis of an RC-1 organic Rankine bottoming cycle system for heavy-duty transport applications. The other major effort was the high-temperature stability testing of the RC-1 organic fluid.

A full description of the work accomplished in the past year is given in the following sections of this report.

2. WORK EFFORTS AND ACCOMPLISHMENTS

Thermo Electron Corporation under contract with NASA Lewis Research Center has participated in their Waste-Heat Utilization Program. During the past year, work efforts were directed towards the following major task areas.

- Thermal Stability Testing of RC-1 Organic Fluid The major goal of these tests is to ascertain the highest operating temperature level of RC-1 through the performance of stability life testing of the organic fluid in a dynamic fluid test loop that simulates the operation of a Rankine-cycle.
- Cycle Analysis This task consists of performing a parametric analysis of a simple organic Rankine bottoming cycle for an adiabatic diesel engine employing a single vapor-generator and RC-1 working fluid. The objective is to identify system design point criteria based on a combination of factors including cycle efficiency, utilization efficiency of the available exhaust gas heat, heat exchanger design, and turbine design. The schedule of exhaust gas conditions versus diesel engine power was provided by NASA Lewis.
- Preliminary System Design Based on the selected design point, this task entailed the preliminary baseline design of the RC-1 organic Rankine bottoming cycle system for the reference 300horsepower diesel engine.
- Life-Cycle Cost Analysis This final task effort is to evaluate the potential capital cost, maintenance cost and, thus, the simple payback and return on investment generated by the fuel savings capability of the baseline bottoming cycle system design defined by the above analysis and design tasks.

2.1 WORKING FLUID STUDIES

The organic fluid RC-1 is a mixture of 60 mole percent pentafluorobenzene (PFB) and 40 mole percent hexafluorobenzene (HFB). Key features of this working fluid for waste-heat utilization from prime movers are:

- Thermally stable at high temperatures
- High chemical stability (resistant to O₂ and H₂O contamination and compatible with materials of construction)

- Excellent thermodynamic characteristics (for power generation from high-temperature gaseous waste-heat sources)
- Nonflammable in air
- Low toxicity (acute and subacute exposures)
- Low freezing point (flow point of -44°F)
- Excellent turbine expansion characteristics (particularly for low power applications)

The initial program work consisted of modifying an existing dynamic fluid test loop to provide capability to test RC-1 organic fluid up to 1000°F for periods up to 1000 hours. Subsequently, the test program was initiated by operating the loop containing the RC-1 fluid at various temperature levels from 700° to 1000°F. In addition to loop testing, static glass capsule tests are being carried out.

2.1.1 Dynamic Loop

The dynamic test loop has been designed to expose the RC-1 working fluid to the conditions that are encountered in operating systems. Samples of the fluid can be chemically examined to determine any degradation that may occur. The idea of the loop is to expose the fluid to very prescribed conditions over fixed periods of time, and chemically measure any changes in the fluid composition.

The loop schematic is presented in Figure 1. The main considerations in the loop design have been: (1) high reliability and leak tightness, (2) loop control for long-term stable operation with a minimum of attention, and (3) key temperature and pressure measurements.

The fluid flow circuit is identical with that of a Rankine power system except that the turbine expander is replaced by a pressure letdown valve. Instrumentation and controls are incorporated to permit unattended round-the-clock operation of the dynamic loop shown in Figure 2.

The fluid loop used for the thermal stability testing of the RC-1 organic working fluid is similar to those used in prior fluid testing at Thermo Electron (i.e., thermal stability testing of Fluorinol). Because the upper temperature limit of RC-1 is higher than ever encountered with organic fluids (greater than 700°F), it was necessary to modify the method of heating the RC-1 to these high temperature levels. Prior testing with Fluorinol utilized a boiler that was electrically heated and used vaporphase heat transfer to the boiler tube carrying the test fluid. Electrical



Figure 1. RC-1 Organic Working Fluid Loop



Figure 2. Front View of RC-1 Dynamic Fluid Loop

heaters were immersed in the heat transfer fluid (Dowtherm A with a temperature limit of 720°F), with the boiler tube located in the vapor space above the heat transfer fluid surface.

This RC-1 loop uses a finned-tube heat exchanger for the vaporgenerator and a forced convection hot air system to heat the RC-1 organic fluid to temperatures up to 1000°F. Figure 3 is a view of the exposed electrical heating elements that heat the air passing over them to the desired temperature levels.

2.1.2 Test Results

More than 1600 hours of testing was completed on the dynamic fluid test loop. Testing was performed at 700°, 800°, and 900°F operating temperature levels. Table 1 summarizes the tests, showing the operating hours and other highlights at each test condition. The three (3) major methods of fluid analysis employed during these tests and their results are:

• Neutralization Number by Color-Indicator Titration (ref. 2)

This process is performed in-house and measures the acidity of the fluid, expressed in milligrams of potassium hydroxide necessary to neutralize one (1) gram of the fluid sample. The allowable acid level established for the former Fluorinol-85 Rankinecycle system was a neutralization number of 0.040. All samples taken in these RC-1 loop tests showed no acid formation. Only the initial sample at about the 90-hour mark of Test No. 1 (700°F) registered a neutralization number of 0.0178, which is well below the 0.040 limit. All other samples for testing performed showed neutralization numbers of zero (0) or very close to it (another sample had a 0.006 neutralization number).

• Gas Chromatography (GC)

A Shimadzu Model 6AM gas chromatograph with an integral thermal conductivity detector was used for this analysis, which is also performed in-house. The samples analyzed during the tests and compared to the pretest stock sample GC of the RC-1 organic fluid show identical constituent peaks in the same proportion. Figures 4 and 5 are GC's of the pretest sample and a sample of the RC-1 fluid at the 606-hour mark of Test No. 2 (800°F operating temperature). Figure 4 is at an attenuation factor of 64 and shows the 606-hour sample to contain the same proportion of RC-1 constituents (PFB and HFB) as the stock working fluid. Figure 5 at an attenuation factor of 8 again shows similar traces with the peak pentafluorochlorobenzene (PFCB) identified with other trace constituents that were already present in the pretest sample.



Figure 3. View of Heating Elements and Hot Air Source for RC-1 Dynamic Fluid Loop

SUMMARY OF RC-1 TESTS IN DYNAMIC FLUID LOOP

	Boiler		Met	hod of Sample Anal	ysis				
Test No.	Outlet Temp. (°F)	Test Hours	Neutralization Number	Gas Chromatography (GC)	Mass Spectral Analysis (MS)	Remarks			
1	700	442	(5) Samples analyzed with no acid for- mation detected	No degradation products de- tected. Traces match pretest RC-1 sample.	Not performed at this oper- ating condition	Test terminated due to rupture of pump diaphragm. Since there was no evidence of thermal degradation at this condition, test was considered complete.			
2	800	653	(5) Samples analyzed with no acid for- mation detected	No degradation products de- tected. Traces at 606 hr match pretest RC-1 sample.	This analysis performed by outside lab with no degradation or changes in RC-1 fluid constituents	All results showed no fluid degrada- tion. Test complete at this condition.			
3	₄ 900	532	(2) Samples analyzed with no acid for- mation detected	Samples ok with proper concen- tration of major fluid constit- uents. Trace peaks observed.	Sample at 417 hours analyzed and shows (2) minor constituent peaks at level of 100 ppm.	Tests conducted through last day of contract period of performance.			

Total 1627

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SAMPLE OF STOCK WORKING FLUID PRIOR TO START OF TEST MIXTURE ANALYSIS: RC-1 (60 MOL % PFB - 40 MOL % HFB) (ATTENUATION × 64)



SAMPLE OF WORKING FLUID AFTER 606 HOURS AT 800°F MIXTURE ANALYSIS: RC-1 (60 MOL % PFB - 40 MOL % HFB) (ATTENUATION x 64)



Figure 4. Gas Chromatographs of Dynamic Loop RC-1 Working Fluid Samples - Test No. 2 at Attenuation x 64

SAMPLE OF STOCK WORKING FLUID PRIOR TO START OF TEST (ATTENUATION x 8) 30 50 20 **4**0 70 80 90 <u></u> 60 ō SAMPLE INJECTION AIR HFB PFB t MIN PFCB SAMPLE OF WORKING FLUID AFTER 606 HOURS AT 800°F (ATTENUATION x 8) 300 8 50 σ 80 œ. 20 ō Ō C О SAMPLE INJECTION AIR HFB PFB 1 MIN PFCB

Figure 5. Gas Chromatographs of Dynamic Loop

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At the 141-hour mark of Test 3 at the operating temperature of 900°F, an RC-1 sample was taken and analyzed. Figures 6 and 7 are the GC's for this sample and also the corresponding pretest sample. Again, Figure 6 shows the proper and similar proportions of the PFB and HFB constituents of the RC-1 fluid. At an attenuation factor of 4 which magnifies the constituent peaks, Figure 7 shows no degradation, but the formation of trace constituents which are extremely minute at the 141-hour mark of the test and not quantitatively determinable. GC's of samples at the conclusion of this 900°F test (532 hours) show similar trace peaks with no major growth to contaminant peaks.

• Mass Spectral Analysis (MS)

This method of analysis provides one of the most stringent and exacting methods for the determination of organic compounds. This analysis method is being carried out by an independent outside testing laboratory. Two (2) samples of RC-1 working fluid were sent out for analysis. One of the samples was taken at the 364-hour mark of Test No. 2 at the operating temperature of 800°F. The second sample was the stock fluid that had been loaded into the loop prior to the test.

The written report (see Appendix A) has been received and the outside laboratory (Cambridge Analytical Associates) indicates that inspection of the Reconstructed Gas Chromatograph (RGC) of the Mass Spectra (MS) yielded identical traces for both samples. In conclusion, no impurities were found in the sample that were not detected in the standard indicating no change in the working fluid at the 800° F temperature after 364 hours.

A gas sample and a liquid sample at the 417-hour mark of the 900°F test were sent to Cambridge Analytical Associates for mass spectral analysis. The gas was analyzed and yielded just one minute peak that is probably a rearranged hexafluorobenzene (HFB) molecule. The analysis of the liquid sample showed no breakdown or changes in the major RC-1 constituents, but did indicate two (2) peaks with molecular weights of 330 and 348. These constitutents were present with concentrations of approximately 100 ppm and are identified as fluoroalkanes. The written report on this 417-hour sample at 900°F operating temperature is presented in Appendix B.

Static Capsule Tests

As a means of further assessing loop results a number of static capsule tests are also being performed. Fluid samples together

314-1083



Figure 6. Gas Chromatographs of Dynamic Loop RC-1 Working Fluid Samples - Test No. 3 at Attenuation x 64

315-1083



Figure 7. Gas Chromatographs of Dynamic Loop RC-1 Working Fluid Samples - Test No. 3 at Attenuation x 4

with metal coupons are placed in glass capsules, sealed, placed in pressurization manifolds (metal containers for the glass capsules), and heated in an oven at various temperatures for a 14day period. Each test series consists of testing a total of 10 capsules containing one (1) milliliter of RC-1 with the following metal coupons:

- 2 tubes with no metal coupons
- 2 tubes with 304 stainless steel coupons
- 2 tubes with 316 stainless steel coupons
- 2 tubes with 1010 carbon steel coupons

- 2 tubes with 6061 aluminum coupons

Static capsule tests were performed at 650°, 800°, and 900°F temperature levels and the results of these tests are presented in Tables 2, 3, and 4, respectively. Table 2, which shows the results of the 650°F test, indicates no change in the metal coupon weights. Physical observations of the metal coupons showed a very slight dulling of the St. St. 304 and Aluminum 6061. The St. St. 316 became slightly browned and the Carbon Steel 1010 turned black. The glass tubes remained clear and the RC-1 fluid water white. GC analysis of the samples indicated no change in the RC-1 fluid.

The 800°F test results are summarized in Table 3 and again show no change in the weights of the metal coupons. All metal coupons were generally blackened as were the glass tubes. The RC-1 fluid turned a yellow brown but no change in fluid constituents occurred as determined by the GC analysis, except for trace peaks with one of the Aluminum 6061 samples.

The 900°F results (Table 4) also show no change in the metal weights except for a 3-percent loss from its original weight in the St. St. 304 coupon. All the metal coupons and glass tubes became blackened. The RC-1 fluid without metal in the capsule was brown in appearance, as was the fluid with the Aluminum 6061 samples. The GC of the Aluminum 6061 fluid had three (3) small, late peaks similar to the Aluminum 6061 sample at 800°F. The fluid from the St. St. 304 samples was yellowish in color with evidence of gas evolution. It had the same chromatographic output as the 900°F Aluminum 6061 samples with the addition of two (2) small peaks on the downside shoulder of the PFB peak. The fluid from the Carbon Steel 1010 capsules was yellowish brown and appeared on the chromatograph with the same three (3) peaks as the Aluminum 6061 samples. The St. St. 316 gave results similar to those of the Carbon Steel 1010.

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SUMMARY OF RESULTS STATIC CAPSULE TESTS AT 650°F

No.	Temp.		Weight (g) Before After			Test	0	bservations	Gas Chromatograph (GC)	
	(°F)	Metal			Condition	(Days)	RC-1 Solution	Metal	Glass	Analysis
1	650 `				Liquid filled Vac < 20 microns	14	Water White		Clear	No Change
2	650				Liquid filled Vac < 20 microns	14	Water White	·	Clear	No Change
3	650	Al 6061	0.1023	0.1026	Liquid filled Vac < 20 microns	14	Water White	Slightly dulled	Clear	No Change
4	650	A1 6061	0.1182	0.1187	Liquid filled Vac < 20 microns	iquid filled 14 No fluid 'ac < 20 microns rupture		Slightly dulled	Clear	No Change
5	650	St. St. 304	0.1679	0.1681	Liquid filled Vac < 20 microns	14	Water White	Very slightly dulled "Silver tarnish" brown	Clear	No Change
6	650	St. St. 304	0.2047	0.2046	Liquid filled Vac < 20 microns	14	Water White	Very slightly dulled, still grey	Clear	No Change
7	650	C.S. 1010	0.1497	0.1476	Liquid filled Vac < 20 microns	14	Water White	Black, still shiny on cut surfaces	Clear	No Change
8	650	C.S. 1010	0.1423	0.1422	Liquid filled Vac < 20 microns	14	Water White	Black, still shiny on cut surfaces	Clear	No Change
9	650	St. St. 316	0.1926	0.1924	Liquid filled Vac < 20 microns	14	Water White	Slightly brownish	Clear	No Change
10	650	St. St.` 316	0.1962	0.1961	Liquid filled Vac < 20 microns	14	Water White	Slightly brownish	Clear	No Change

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SUMMARY OF RESULTS STATIC CAPSULE TESTS AT 800°F

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No.	Temp.	Metal	Weight (g)		Condition	Test Time		Observations	Gas Chromatograph (GC)			
	(°F)		Before	After	Condition	(Days)	(Days) RC-1 Solution		Glass	Analysis		
1	800				Liquid filled Vac < 20 microns	14	Brown	wn Black		No Change		
2	800				Liquid filled Vac < 20 microns	14	Brown	·	Black	No Change		
3	800	A1 6061	0.0975	0.0983	Liquid filled Vac < 20 microns	14	Brown	Black	Black	No Change		
4	800	Al 6061	0.0726	0.0733	Liquid filled Vac < 20 microns	14	14 Brown Black Black		Black	Few small late peaks		
5	800	St. St. 304	0.2418	0.2417	Liquid filled Vac < 20 microns	14	Tube ruptured	Tube ruptured Brown-Blue Tube ruptu		No Change		
6	800	St. St. 304	0.2444	0.2444	Liquid filled Vac < 20 microns	14	Brown	Brown Black Black		No Change		
7	800	C.S. 1010	0.1911	0.1917	Liquid filled Vac < 20 microns	14	Brown	Black	Black	No Change		
8	800	C.S. 1010	0.1662	0.1667	Liquid filled Vac < 20 microns	14	Brown	Black ,	Black	No Change		
9	800	St. St. 316	0.2074	0.2077	Liquid filled Vac < 20 microns	14	Brown	Black	Black	No Change		
10	800	St. St. 316	0.1939	0.1940	Liquid filled Vac < 20 microns	14	Brown	Black	Black	No Change		

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SUMMARY OF RESULTS STATIC CAPSULE TESTS AT 900°F

No.	Temp.	Metal	Weight (g)		Condition	Test Time	Observ	ations		Gas Chromatograph (GC)		
	(°F)	etc.	Before	After	Condition	(Days) R		Metal	Glass	Analysis		
1	900				Liquid filled Vac < 20 microns	14	Tube ruptured					
2	900				Liquid filled Vac < 20 microns	14	Very little and brown		Black	Not enough to analyze		
3	900	Al 6061	0.0765	0.0832	Liquid filled Vac < 20 microns	14	Brown	Black	Black	Formed three, small, late peaks		
4	900	Al 6061	0.1063	0.1129	Liquid filled Vac < 20 microns	14	Brown	Black	Black	Formed three, small, late peaks		
5	900	St. St. 304	0.2141	0.2072	Liquid filled Vac < 20 microns	14	Yellow gas evolved	Black	Black	Formed small peaks on shoulder of PFB peak		
6	900	St. St. 304	0.1859	0.179	Liquid filled Vac < 20 microns	14	Yellow gas evolved	Black	Black	Formed small peaks on shoulder of PFB peak		
7	900	C.S. 1010	0.1557	0.1578	Liquid filled Vac < 20 microns	14	Yellow brown	Black	Black	Formed three, small, late peaks		
8	900	C.S. 1010	0.1486	0.1499	Liquid filled Vac < 20 microns	14	Yellow brown	Black	Black	Formed three, small, late peaks		
9	900	St. St. 316	0.2189	0.2193	Liquid filled Vac < 20 microns	14	Yellow brown	v brown Black Blac		Formed three, small, late peaks		
10	900	St. St. 316	0.2212	0.2214	Liquid filled Vac < 20 microns	14	Yellow brown	Black Black		Formed three, small, late peaks		

GC's of the stock RC-1 sample prior to the tests and GC's of all the capsule samples at the conclusion of the static tests were taken for all RC-1 fluid-metal combinations and for all temperature conditions (650° , 800° , and 900° F).

2.2 SYSTEM ANALYSIS

The initial efforts of this task consisted of the acquisition of the RC-1 thermodynamic properties from Monsanto with subsequent storage and formating of the tables for use in Thermo Electron's computer programs. The thermodynamic properties of RC-1 are tabulated in Appendix C.

Studies were made of the effect on performance by varying system component sizes and other variables. Many iterations and computations were undertaken to optimize the system before the design point was chosen.

Data on an adiabatic engine in four (4) different configurations were supplied by NASA Lewis, and the performance of an RC-1 Rankine bottoming cycle system was computed for these engines.

2.2.1 RC-1 Parametric Analysis

To determine the most efficient RC-1 operating cycle for this application, a parametric analysis was undertaken. By varying the system operating pressures and temperatures as well as the component efficiencies, it was possible to determine these effects on the overall system performance and the sizes required for the system's heat exchangers. Using this information, a system design point, including the cycle state points and component designs, could be specified.

The parametric analysis proceeded by first identifying a "baseline cycle" state point condition. Then each parameter, whose effect on the system's performance was to be measured, was varied (in turn) and the results of this change to the Rankine-cycle operating conditions were recorded. The baseline condition selected for this RC-1 parametric study is shown in Table 5. Also identified in Table 5 is the range of exhaust gas temperatures of interest as well as the imposed limits of turbine inlet temperature and pressure used in the study. The desired result of any cycle calculation was: the overall cycle conversion efficiency, the identification of component sizes, and the efficiency of the component. The conversion efficiency is not the same as Rankine-cycle efficiency. Conversion efficiency is defined as the ratio of the net cycle power to the maximum power available in the exhaust referenced to 300°F. The imposed

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TABLE 5RC1 PARAMETRIC ANALYSIS

Baseline Case:

 $\Delta P_{VG} = 70 \text{ psid}$ $\Delta P_{Cond.} = 0.5 \text{ psid}$ $\Delta P_{Regn. Liq.} = 10 \text{ psid}$ $\Delta P_{Regn. Vap.} = 0.5 \text{ psid}$ $\Delta T_{VG} (Min.) = 35^{\circ}F$ Regenerator Eff. = 90%
Pump Eff. = 50% $T_{Cond.} = 170^{\circ}F$ Turbine Thermal Eff. = 75%
Turbine Mech. Eff. = 95% $1000^{\circ}F \leq T \leq 1400^{\circ}F$ $600^{\circ}F \leq T \leq 900^{\circ}F$ $300^{\circ}F \leq p \leq 1000 \text{ psia}$

^TExh. Gas Range[†] ^TTurbine Inlet Range[†] ^PTurbine Inlet Range[†] ^PCritical = 411 psia ^TCritical = 456°F

Exhaust Gas Flow Rate = 4000 lb/hr

$$\left(C_{P \text{ Gas}} = 0.26 \frac{Btu}{lbm^{-} \circ F} \right)^{2}$$

^TExhaust Min. Stack Temp. = $300^{\circ}F$ Conversion Eff. = $\frac{bhp}{m \times Cp \times (T_{Exh}, In^{-300^{\circ}F})}$ limit of 300°F was selected as the minimum exhaust temperature, below which particulates could condense out of the exhaust stream to cause corrosion problems on the heat exchanger surfaces. The conversion efficiency equation is defined at the bottom of Table 5.

Five (5) different cases were studied in this parametric analysis. Each case studied the effect of changes of one parameter on the overall cycle performance. A complete list of the parametric variations performed in each case is shown in Table 6. These five (5) limited cases by no means represent all the parametric variations that could be considered in a very detailed Rankine-cycle study. However, they do represent the major parametric variables that have the most effect on the selection of a design point for a waste-heat recovery system. For example, a parametric study could also involve verifying the pressure drops of the system's components. In fact, if the pressure drops were varied from zero (i.e., no pressure drop at all) to the values shown in Table 5, a variation of only 1 to 3 percent in overall conversion efficiency would result. A full summary of the effect of component drop on conversion efficiency is given in Table 7.

The results of each case study were arranged in both graphical and tabular form. This arrangement provided the best representation of the data and facilitated the observation of component size or system performance trends. The graphical results for the Case I study are presented in Figures 8 through 11. By referring to these figures it is possible to quickly determine the conversion efficiency as a function of RC-1 system operating temperature and pressure and exhaust gas inlet temperature. Of equal importance is the effect of these parameter changes on the overall sizes of the heat exchanger equipment needed to obtain that particular system performance. By utilizing this graphical representation of the results, the trends of increasing or decreasing component sizes become more readily apparent.

The tabular data compiled for Cases I, II, III, and V are presented in Tables 8, 9, 10, and 11, respectively. The study data were tabulated as shown in these tables to quickly quantify the effect of changes in the various parameters. By observing Table 8 (parameter summary for Case I), it is possible to quickly determine the magnitude of change in the conversion efficiency as a function of the RC-1 operating temperature and pressure and exhaust temperature.

From Table 8 it is also possible to quantify the effect of this parametric variation on the condenser, regenerator, and vapor-generator. Similarly, Tables 9, 10, and 11 summarize the parametric studies for Cases II, III, and V.*

*Case IV is the effect of turbine efficiency and all the tables and figures shown here were repeated for a tubine efficiency of 0.65.

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TABLE 6

PARAMETRIC STUDY VARIATIONS

CASE I: BASELINE CASE

WITH P_{TURBINE} INLET = 300, 400, 500, 700, 1000 PSIA

AND T_{EXH} . GAS = 1000°, 1200°, 1400°F

CASE II: VARIATION OF CONDENSING TEMPERATURE

 $T_{COND.} = 140^{\circ}, 160^{\circ}F$

CASE III: VARIATION OF REGENERATION TEMPERATURE

 $N_{REGN.} = 0, 0.5, 0.75$

CASE IV: VARIATION OF TURBINE EFFICIENCY

 $N_{TN} = 0.65, 0.75$

CASE V: VARIATION IN MINIMUM APPROACH TEMPERATURE IN VAPOR GENERATOR

 $\Delta T_{MIN} = 70^{\circ}, 100^{\circ}F$

IABLE /	Т	A	В	LE	7
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EFFECT OF COMPONENT PRESSURE DROP ON CONVERSION EFFICIENCY

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Exhaust Temperature	1000°F									1200°F								1400°F								
Turbine Pressure	300 400		500		1000 300		00	40	10	50	0	10	00	30	0	4(00	50	0	10	00					
Turbine Temperature	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900		
ⁿ conv. ⁿ conv. ideal	0.97	0.99	0.97	0.98	0.97	0.96	0.97	0.97	0.97	0.99	0.96	0.98	0.97	×	0.97	0.97	0.96	0.99	0.97	0.99	0.97	×	0.96	0.97		

 $n_{conv.}$ = Conversion Efficiency with Component Pressure Drops Shown for Case I

 $\eta_{conv.}$ ideal = Conversion Efficiency with Component Pressure Drops Equal to Zero (0)

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Figure 9. Case I - Graphical Results of RC-1 Pressure (300 psia) and Exhaust Temperature on Heat Exchanger Sizes





Figure 10. Case I - Graphical Results of RC-1 Pressure (500 psia) and Exhaust Temperature on Heat Exchanger Sizes





Figure 11. Case I - Graphical Results of RC-1 Pressure (1000 psia) and Exhaust Temperature on Heat Exchanger Sizes
TABLE 8

CASE I - BASELINE PERFORMANCE

Exhaust Temperature In				100	0°F							120	0°F							140	0°F			
Turbine Pressure	3	00	41	00	. 50	00	10	00	30	00	41	00	5	00	10	00	3(00	41	00	51	00	10	00
Turbine Temperature	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900
$\frac{(\eta_{conv.})}{(\eta_{conv.})}$	1	0.87	1.047	0.967	1.095	1.041	1.151	1.098	1.05	0.938	1.101	1.047	1.148	1.145	1.192	1.207	1.089	0.979	1.136	1.101	1.183	1.213	1.219	1.278
UA regen.	1	0.708		ł	1.011	0.775	1.275	0.850	1.364				1.421	1.108	1.683	1.22	1.45	1.225			1.80	1.300	1.838	1.325
(UA/HP) regen.	83.1	66.9			79.7	62.2	90.6	65.0	83.0	66.9			79.6	62.3	90.6	65.0	83.1	66.9			79.7	62.3	90.6	65.1
UA cond.	1	0.738	[1.048	0.810	1.143	0.810	1.464	1.054			1.474	1.166	1.541	1.155	1.76	1.333			1.81	1,667	1.905	1.476
(UA/HP) cond.	468.0	379.8		ł	433.8	343.7	435.5	322.4	467.9	379.5			433.5	343.9	435.5	322.5	468.4	379.8	ļ		433.8	343.9	435.5	322.6
UAVE	1	1.125			1.063	2.08	1.10	1.6	1.013	0.765	}		1.054	1.483	1.072	1.183	1.038	0.65			1.075	1.188	1.075	1.100
(UA/HP)vg	83.0	107.9	ļ		80.8	165.1	78.3	120.5	61.7	52.5			59.1	83.3	57.7	62.9	49.9	35.6			47.4		46.5	45.0

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CASE II - EFFECT OF CONDENSER TEMPERATURE ON TOTAL PERFORMANCE

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Exhaust Gas Temperature				10	000°F						·	1	200°F							1	400°F			
Turbine Pressure	3	00	4(00	. 50	00	10	00	3	00	40	0	5	00	10	00	3	00	4(00	5	00	10	000
Turbine Temperature	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900	700	900
$UA_{cond.}$ $(T = 170^{\circ}F)$	1	0.738			1.048	0.81	1.143	0.810	1.46	1.054			1.474	1.166	1.541	1.155	1.76	1.333			1.81	1.667	1.905	1.476
(UA/HP)	468	380			434	344	436	322									468	380			434	344	436	323
UA cond. (T_=160°F)	1.72	1.19			1.75	1.28	1.86	1.28									2.9	2.1			Z. 96	2.34	3.1	2.34
C (UA/HP)	724	573	Ň		674	523	683	494									723	573			675	522	683	494
n _{conv.}	1.03	0.90			1.12	1.07	1.18	1.13									1.12	1.02			1.21	1.24	1.25	1.31
UA cond. (T = 140°F)	1.37	0.997			1.39	1.05	1.47	1.09									2.32	1.76	i i		2.33	1.9	2.43	1.93
(UA/HP)	529	432			497	396	501	375									530	431			497	396	501	375
nconv. nconv.,o	1.12	1			1.22	1.15	1.28	1.26									1.21	1.13			1.3	1.34	1.34	1.42

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CASE III - EFFECT OF REGENERATOR EFFICIENCY ON TOTAL PERFORMANCE

Exhaust Gas Temperature			10	00°F					140	0°F		
Turbine Pressure	30	0	50	0	. 10	00	3(00	5()0	10	00
Turbine Temperature	700	900	700	900	700	900	700	900	700	900	700	900
at Regn. Eff. = 0.90												·
$\eta_{\rm conv}, \eta_{\rm conv0}$	1	0.87	1.1	1.04	1.15	1.1	1.1	0.98	1.18	1.21	1.22	1.28
UA regn. /UA regn0	1	0.71	1.01	0.78	1.28	0.85	1.45	1.23	1.80	1.30	1.84	1.33
(UA/HP)	83	67	80	62	91	65	83	67	80	62	91	65
at Regn. Eff. = 0.75 ⁿ conv. ^{/n} conv.,o UA _{regn.} ^{/UA} regn.,o (UA/HP)	0.953 0.50 43.2		1.05 0.51 40.6	0.98 0.39 32.8	1.11 0.58 42.9	1.07 0.41 31.9	1.02 0.83 43.3	-	1.11 0.85 40.5	1.11 0.69 32.9	1.16 0.94 42.8	1.20 0.72 31.9
at Regn. Eff. = 0.50		0.017	0.00		1.07	1 01	0.02	0.00	1 02	0.08	1 09	1 07
$\eta_{\text{conv.}}/\eta_{\text{conv.},o}$	0.91	0.817	0.99	0.91	1.07	1.01	0.93	0.00	0.33	0.70	0.36	0.28
regn. regn.,o	0.20	0.10	0.61	0.10	17 6	12.0	18 5	16 2	17 1	14 6	17.6	13.8
(UA/HE)	10.5	10.1	11.6	14.0	11.0	17.0	10.0	10.2	11.1	11.0		1310
UA/UA Vapor. Gen.		1 125		. 2 1	1 1	1.6	1 04	0.65	1 08	1 19	1.08	1.10
regn. = 0.75		1.140	1.00	1 85	1.1	1.0	1 10	1 24	1.1	1.24	1.12	1.22
at $\eta_{regn.} = 0.75$ at $\eta_{regn.} = 0.50$	1.10	1.50	1.14	1.83	1.26	1.75	1.19	1.24	1.19	1.29	1.18	1.29

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CASE V - EFFECT OF VAPOR GENERATOR PINCH POINT TEMPERATURE ON TOTAL PERFORMANCE

Exhaust Temperature			1000)°F					14(00°F		
Turbine Pressure	3	00	5	00	10	00	3(00	5	00	10	00
Turbine Temperature	700	900	700	900	700	900	700	900	700	900	700	900
$\Delta T_{vg} \approx 35$ UA/UA _{vg,0} ⁿ conv. ^{/n} cony.,0 UA/HP	1 1 83	1.125 0.87 108	1.06 1.10 81	2.08 1.04 165	1.1 1.15 78	1.6 1.1 121	1.04 1.09 50	0.65 0.98 36	1.08 1.18 47	1.19 1.21	1.08 1.22 47	1.1 1.28 45
$\Delta T_{vg} = 70$ $UA/UA_{vg,o}$ $\eta_{conv.}/\eta_{conv.,o}$ UA/HP	0.75 0.94 66	0.93 0.817 94.6	0.80 1.03 64.6	1.40 0.96 119.9	0.82 1.08 62.7	1.19 1.02 96.8	0.81 1.05 40.7	0.60 0.94 33.3	0.84 1.14 38.9	1.02 1.166 46.1	0.85 1.18 38.2	0.90 1.23 38.5
UA/UA vg,o n _{conv.} /n _{conv.,o} UA/HP	0.62 0.89 58.0	0.79 0.76 86.3	0.67 0.97 56.9	1.09 0.90 100.4	0.68 1.03 55.2	0.97 0.95 84.2	0.70 1.02 36.3	0.55 0.91 31.6	0.73 1.10 34.7	0.86 1.12 40.4	0.74 1.14 34.1	0.78 1.18 34.7

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The detailed study and evaluation of all the tabular and graphical results compiled from this parametric study led to the selection of the system design point. This selection, although not necessarily providing the highest conversion efficiency, does consider the tangible consequences of component size and packaging requirements for installation as an integrated system on a heavy-duty vehicle. A detailed state point summary of the design point selected is shown in Figure 12. These particular results and system operating conditions consider the use of a water-cooled condenser. An improvement in performance can be demonstrated by the use of a direct air-cooled condenser. The design point results and conditions for the air-cooled system are shown in Figure 13. Based upon this improved performance and the elimination of water cooling loops in the Bottoming Cycle System, as is accomplished with the adiabatic diesel engine, the RC-1 system utilizing the air-cooled condenser was selected for final component sizing and specification.

2.2.2 Diesel/RC-1 Bottoming Cycle Power Summary

The baseline diesel engine data supplied by NASA Lewis are shown in Table 12. Four (4) diesel configurations with performance and exhaust gas conditions were provided for the waste-heat recovery performance evaluation.

Using the waste-heat energy from these base diesel engines, an RC-1 bottoming cycle system performance was calculated for each diesel engine configuration. Performance was determined for RC-1 systems using both the water-cooled and air-cooled condenser options. The operating state point conditions used in this analysis were those determined to be optimum from the parametric study. A typical RC-1 state point and thermodynamic process diagram is shown on the temperature-entropy plot in Figure 14. Table 13 contains the performance data for the RC-1 bottoming cycle system matched to each diesel engine configuration. Bottoming cycle power as high as 56 horsepower for the air-cooled condenser option and 53 horsepower for the water-cooled condenser option are possible. In all instances, the performance calculations did not consider engine fan power parasitics. This rationale is consistent with the performance presentation of the base diesel engines given in Table 12.

The compound engine (diesel plus bottoming cycle) performance is shown in Table 14. The analysis indicates that a minimum compound engine $\Box S^{1} \cup$ of 0.258 lb/hp-hr can be obtained if an RC-1 Bottoming Cycle System (with the air-cooled condenser option) is used to bottom the turbocharged, turbocompounded air after cooled diesel engine (TCPD/A). For the simple turbocharged diesel engine (TC), a maximum fuel savings and power improvement of 15 and 18 percent, respectively, is possible with the RC-1 Bottoming Cycle System.

		r	Ч	н		זע	90	ЫН
14	11 JI08 100	327.65 750.00	370.00 300.00	82.25 232.67	BUIL	422.352 .000	29.000 .000	150.423 .000
3	ENG IN OUT	750.00 551.58	300.00 10.09	232.67 195.86	eng Line	198.423 .000	789.910 .000	36.915 .000
4 5	NI UDIA TUC	551.58 331.16	10.09 9.59	195.86 145.86	REGU L INE	220.419 .000	. 500 . 000	49.996 .000
6 9	אד מאנ טט ז'נוט	331.16 148.25	9.59 9.09	145.86 28.55	COND LINE	182.910 .000	. 500 . 800	117.314 .000
19 11	MI am uq Tuq	148.25 161.80	9.09 880.00	28.55 32.25	PUMP LINE	13.547 .000	-870.910 .900	3.705 .009
12 13	R ECL IN OUT	161.80 327.65	339.09 379.09	32.25 82.25	HEGL LINE	165.853	10.000 .000	4 9. 996 . 600
SYSTEM	1						•	
FLOW	IRA'TE =	4401.	= 4H	52.55	:	EFF	= 20.20	
EXPANI	DER .	= لَرَا	. 162	DE DE	HP =	58.96		
EFFI	H = .7651	а	EFFME =	. 9260		EFFH	LL = .70	84
WES	= 48,12		WSHAFT	= 34.09	1	UNET	= 39.	39
EXHF	NUSTFOUNL I	ſY ≈ 1.000						
RECENE	RHITOR	EFF	= .608	60	Ŭ =	.2201E	96	
UA =	1126.							
CUNDEN	ISER	(j) =	.516	4E US		บลาง	TAL = 42	771.
HSUE	3 = 1.4	41	HUESUP	= 33.8	4	HLHT	ENT = 8	2.06
UASU	0 8 = 55	.98.	ансони	= 3929	ช.	UHDE	SUP =	2881.
TMAT	тер IN = 3	140.0	NHATER	UUT = 1	50.0	NATE	rflow =	51635.
BUILER	•	ψ =	. 662	1E 06		UHDD	[AL = .3	635.
HSEN	IS = - 59,75	5	HLATENT	'= .u	ហ	HBUE	EK = 90.	67
TBSU	₽ = 178.5	91						
UASE	1 18 = 25	520.	nurùi =	ن . ن	•	UHSU	PER = 1	115.
AW NA								
· EFF	= .5000		GPM =	6.47		HP =	÷.,	41 .
GHS								
FLOH	KHILE =	2886	TGHS1 =	1240.0		TGHS	j = .352.	. ύ
TPIN	CH =	3.1	DIFINCH	= <u></u> 30	.ΰ			

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Figure 12. Computer Output of Design Point for Water-Cooled RC-1 System

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14 EOIL IN 1 OUT	T 316.76 750.00	P 870.00 800.00	H 78.59 232.67	BOTL LINE	01 433.236 .000	09 70.090 .000	08 154.081 .999
2 ENG IN 3 007	$\times 750.00$ 541.79	800.00 8.01	232.67 193.52	ENG LINE	208.212 .600	791.990 .000	39.147.000
4 REGU IN 5 OUT	541.79 020.26	8.01 7.51	193.52 143.66	REGU LINE	221.528 .000	.500 .000	49.868
6 COND IN 9 OUT	320.26 135.00	$7.51 \\ 7.61$	143.66 25.08	COND L INE	185.260 .000	.500 .000	118.570 .000
10 PUMP IN 11 OUT	135.00 148.62	7.01 880.00	25.08 28.72	FUMP L IME	13.615 .000	-872.990 .000	3.635 .000
12 REGL IN 13 OUT	148.62 316.76	88 0.00 870.00	28.72 78.59	REGL L INE	168.149 .000	10.000 .000	49.868 ,000
SYSTEM							
FLOURATE =	4350.	HF, =	55.75		EFF	= 21.17	,
EXPRNDER	0 =		3E 86	HP =	61.96	,	
EFFTH = .76	50	EFFME =	. 9260		EFFF	LL = .70	184
MES ≈ 51.17		изнает	= 36.25		UNET	= 32.	61
EXHAUST QUAL	ITY = 1.000						
REGENERATOR	· EFF	600	ម	Q =	.2169E	06	
· UA = 1100.							
CONDENSER	0 =	.515	8E 06		URTO	TAL = 9	970.
HSUB ± 1	.34	HDESUP	= 33.7	6	HLH	ENT = 8	3.48
UHSUB =	101.	URCOND	= 825	8.	UADE	SUP =	1611.
TUATER IN =	80.0	TURTER	OUT = 1	20.0	инте	RFLOW =	53727.
BOILER	() =	.678	2E 06		UPPTC)7AL = 3	691.
HSENS = .	00	HLATENT	= .0	Ø	HSUF	ΈR = .	00
TBSUP = 178	.91		· ·.				
UASENS =	2601.	UALAT =	Ū.		UASU	IPER = 1	090.
PUMP							
EFF ≈ .5000		GPM =	6.25		₩ ₽ * =	÷.	21
GAS							
FLOWRATE =	2886	TGASI =	1240.0		TGAS	Ú = 346	.8
TPINCH = 7	14.4	DIFINCH	= 30	.0			

Figure 13. Computer Output Design Point for Air-Cooled RC-1 System 36

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TABLE 12 DIESEL ENGINE DATA

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DIESEL	внр	BSFC (LB/BHP-HR)	EXHAUST (°F)	EXHAUST (LB/MIN)
TURBOCHARGED-NONAFTERCOOLED (TC)	317	0.315	1240	48.1
TURBOCHARGED-AFTERCOOLED (TC/A)	320	0.310	1120	47.6
TURBOCOMPOUND-NONAFTERCOOLED (TCPD)	335	0.297	1140	47.8
TURBOCOMPOUND-AFTERCOOLED (TCPD/A)	340	0.293	1060	48.4





Figure 14. RC-1 State Point Diagram for Design Point Thermodynamic Process

TABLE 13 BOTTOMING CYCLE DATA

	TUR EFFIC	BINE	w	ATER-COOL	ED CO	NDENSER		AIR-COOLE	D CON	DENSER
DIESEL CONFIGURATION	ηth ⁽¹⁾	, (2) η _{ΟΑ}	HCYCLE ⁽³⁾	(4) ⁿ conv.	внр	EXHÂUST TEMPERATURE ⁽⁵⁾ (°F)	(3) ⁿ cycle	¹ CONV.	внр	EXHAUST TEMPERATURE ⁽⁵⁾ (°F)
тс	0.765	0.708	0.202	0.19	52.6	358	0.212	0.201	55.8	347
TC/A	0.788	0.728	0.211	0.198	47.5	349	0.229	0.215	51.5	348
ТСРД	0.765	0.706	0.201	0.187	46.1	358	0.218	0.204	50.3	352
TCPD/A	0.765	0.708	0.197	0.188	42.5	334	0.222	0.206	46.4	355

NOTES:

- (1) $n_{EXPANDER} = \frac{TURBINE THERMAL POWER OUTPUT}{ISENTROPIC IDEAL POWER} \frac{(\Delta H)}{(\Delta H_s)}$
- (2) TGU = TURBINE GEARBOX UNIT OVERALL EFFICIENCY
- (3) ⁿCYCLE = $\frac{BHP}{M \times c_p \times (\Delta T)}$; $c_p = 0.26 BTU/LBM^{\circ}F$
- (4) $n_{\text{CONVERSION}} = \frac{BHP}{fm \times c_p \times (T_{\text{GAS IN}} 300)}$
- (5) EXHAUST GAS STACK TEMPERATURE (OUT OF VAPOR GENERATOR)

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TABLE 14

DIFEE		WATER-COOLE	D CONDENS	ER		AIR-COOLED	CONDENSER		AHP (3)
CONFIGURATION	внр	BSFC (LB/HP-HR)	∆BSFC ⁽¹⁾ (%)	∆HP ⁽²⁾ (%)	внр	BSFC (LB/HP-HR)	∆BSFC ⁽¹⁾ (%)	∆HP ⁽²⁾ (%)	500 FL85 (%)
тс	369.6	0.270	14.2	16.6	372.8	0.268	14.9	17.6	14.8
TC/A	367.5	0.270	12.9	14.8	371.5	0.267	13.9	16.1	13.0
TCPD	381.1	0.261	12.1	13.8	385.3	0.258	13.1	15.0	12.8
TCPD/A	382.5	0.260	11.1	12.5	386.4	0.258	11.9	13.6	11.6

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COMPOUND ENGINE SYSTEM DATA

NOTES:

(1)
$$\Delta BSFC$$
 (%) = $\left(1 - \frac{(BSFC)_{COMPOUND}}{(BSFC)_{WITHOUT COMPOUND}}\right) \times 100$
(2) ΔHP (%) = $\left(\frac{HP_{COMPOUND}}{HP_{WITHOUT COMPOUND}} - 1\right) \times 100$
(3) ΔHP_{FL85} (%) = $\left(\frac{HP_{FL85 ORCS}}{HP_{WITHOUT COMPOUND}} - 1\right) \times 100$

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An analysis to determine the performance sensitivity with respect to exhaust gas temperature was also conducted, and the results of this study are shown in Table 15. The table shows that a compound engine BSFC as low as 0.251 lb/hp-hr is possible if the exhaust temperature from the adiabatic diesel engine reaches 1600° F. This represents a fuel savings potential of 19 percent and a bottoming cycle power delivery of 75 horsepower.

2.3 PRELIMINARY SYSTEM DESIGN

All of the power performance calculations presented in the previous section for the diesel bottoming cycle system compound engine were performed with fixed Organic Rankine-Cycle System (ORCS) component sizes. These component sizes were chosen because they could be realistically packaged and installed in a heavy-duty transport vehicle equipped with an adiabatic diesel engine.

Thermo Electron's prior experience with the Fluorinol-based truck bottoming cycle system was of considerable advantage in the design of each component for the RC-1 system. These advantages also included the general packaging arrangement for an effective vehicle installation. Because of the properties of the RC-1 organic fluid, it was necessary to redesign or modify the four (4) principal Rankine-cycle components: the vapor-generator, feedpump, turbine-gearbox unit, and integrated aircooled condenser-regenerator heat exchangers. The bottoming system schematic showing the relative location of each component and the operating conditions for this RC-1 system at the design point is shown in Figure 15. A description of the major components and their design features are presented below.

Vapor-Generator

The vapor-generator for this system has a core area which is only 11 percent larger than the one for the Fluorinol-85 truck ORCS. For the same diameter unit, this translates into a 4-inch increase in the overall length of the finned-tube core. However, at this increased length the core will fit inside the 48-inch-long vapor-generator shroud, thus maintaining the same package size for the vapor-generator module. The core consists of finned tubing spirally wound in "pancake" sections as in the VGIII design for the Fluorinol system. All other features, such as the diverter valve and internal bypass stack, also remain the same. Figure 16 shows the basic design of the vapor-generator.

• Feedpump Design

The RC-1 feedpump design is identical with the Fluorinol truck system design except for an increase in the cylinder displacement.

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TABLE 15

TEMPERATURE SENSITIVITY FOR TC/A DIESEL CONFIGURATION

	WATER	-COOLED CONI	DENSER	AIR-	COOLED CONDE	ENSER
TEMPERATURE (°F)	ORCS (BHP)	COMPOUND BSFC (LB/HP-HR)	∆BSFC (%)	ORCS (BHP)	COMPOUND BSFC (LB/HP-HR)	∆BSFC (%)
1000	38.9	0.276	11.0	44.3	0.272	12.3
1200	50.8	0.268	13.5	56.1	0.264	14.8
1400	60.7	0.261	15.8	66.6	0.257	17.1
1600	70.7	0.254	18.1	75.2	0.251	19.0

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*BASELINE DIESEL (TC/A)

BHP = 320

BSFC = 0.310 LB/HP-HR EXHAUST TEMPERATURE = 1120°F EXHAUST FLOW RATE = 2856 LB/HR







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Figure 16. RC-1 Vapor Generator Design

25-1082

Basically, it is a three (3) cylinder, variable positive displacement pump with a flow rate capability from 0 to $6\frac{1}{2}$ gpm at 1000 psia design pressure. It has a cylinder base of 1.1 inches and a maximum stroke of 0.315 inch.

The change in pump displacement is accomplished by using a small electric screw-motor that axially moves a Z-bar camshaft which changes the eccentricity of a rotating cam follower. The net effect is the capability to vary the piston stroke from 0 to 0.315 inch.

The pump is integrated with a feedpump drive unit and installed as a package as part of the PCU.

Turbine-Gearbox Unit

The RC-1 turbine and gearbox requirements were reviewed very carefully. The RC-1 turbine operating conditions were sufficiently different from the Fluorinol-85 conditions to warrant a new turbine design analysis. Using several organic fluid-turbine design computer programs and inputting the new RC-1 operating conditions (i.e., 750°F and 800 psia turbine inlet conditions), certain turbine specifications were identified that were used in finalizing a turbine design. The turbine design is defined in the computer output of Figure 17 and shown in Figure 18. The turbine rotor is $3\frac{1}{2}$ inches in diameter and will turn at a speed of 55,000 rpm. The shaft seal is a double-faced carbon seal with a single mating ring and a buffer oil zone between the seals to prevent leakage of RC-1 fluid out of the system and at the same time prevent air leakage into the subatmospheric system. Hydraulic bearings are used in this design instead of roller bearing elements, and the turbine housing and rotor are expected to be cast for cost effectiveness.

The turbine gearbox reduces the speed of the turbine (55,000 rpm) to 3000 rpm at the Power Take Off (PTO) input gear. This speed reduction is accomplished by a planetary gear train that is contained in a small compact housing. The turbine output power is transmitted to the diesel driveline through an over-running (Sprague-type) or one-way clutch. Thus, the turbine-gearbox unit is not a parasitic and cannot be turned by the diesel engine. Only when the Bottoming Cycle System is putting out power will the clutch engage and deliver positive power to the diesel driveline.

The turbine, gearbox, feedpump, and feedpump drive assembly are installed on the PTO as a single assembly. There is a single

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ISTART \$ 2800., 750., 4408., 10.09, 10., 10000., 3000., 1 STEAM CALC.S TRUE OR FALSE ?.FALSE. P1 = 800.000 71 = 750.000 FLOW = 4408.00 PE = 10.0900 UERSP = 10.0000 RLIFE = 10000.0 ARPM = 3000.00 INPUT NTNE (FIRST STAGE TO BE RUN) 21 INPUT NTNL (LAST STAGE TO BE RUN) ?1 INPUT NOPT=1 FOR OPT. RUNS...NOPT=0 IF NO 20 1STAGES STAGE NUMBERS З ទ 2 1 4 PIN 800.00 TIN 749.37 PEST 9.84 HSTRICE OVERALL PERFORMANCE GBR (1-7) 48.42 STNS .1837**2E** 02 GBN 56.26 SD .30000E 01 1.46 SHAFT .44017E 00 D THRUST 3.47 .12031E 01 BHE IGHT ROLML .53 URM .25120E 00 .50 BRGNO TIPUEL .20200E 03 835.67 BRGSHHFT REACTION .59055E 00 BL IFE .00 UU .17213E 06 779.83 AUGLIFE HLPHR2 .36067E 06 16.00 SEALD .890**5**5E 00 SEALSP BE LUS 31.54 .21417E 03 SEALHP W2 802.21 REFWCS .72790E 00 THEFF 1.76 ALPHA3 .76503E 00 90.00 HPITHERM BETA3 .63717E 02 15.20 GBEF .94929E 00 MECHEF ΜЗ 808.09 RELMC3 .97572E 00 TGUEF 1.77 PTHH .70861E 00 498.23 TRPMM **ULHH** .55117E 05 . 12 TGUHPNET HHORK .59017E 02 *STOP* 0 36.82 TTE . 86 STE

Figure 17. RC-1 Turbine Design Computer Output

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Figure 18. Turbine Design

mechanical interface between the RC-1 bottoming cycle PCU and the vehicle PTO. With this new arrangement, only one (1) gear in the PTO must be engaged, which not only reduces the complexity but also allows the bottoming cycle system to be more universally applied to various engine systems. The PCU module containing the turbine, gearbox, and feedpump assembly is shown in Figure 19.

Condenser-Regenerator Module

The integrated air-cooled condenser-regenerator is the most unique subsystem in the RC-1 bottoming cycle system design. The existing Fluorinol-85 Truck Bottoming Cycle System consists of a water-cooled condenser and an air-cooled radiator heat exchanger cooling loop. With the arrival of the adiabatic engine, there is now considerable incentive to eliminate any and all radiators in front of the diesel engine. The air-cooled condenser design for this system eliminates the need for a water-cooling loop and the module is located to the rear of the engine and vehicle cab (as is the vapor-generator). The regenerator in this design is integrated into the top header of the condenser. This arrangement eliminates the pipe connections that would otherwise be required of two (2) separate heat exchangers and also minimizes the installation space required for the heat exchanger. The condenser-regenerator module, shown in Figure 20, consists of a unique condenser design. The air-cooled condenser is cylindrically shaped permitting cooling air to flow radially across the condenser from the inside to the outside diameter surface. The heated air is then rejected to the ambient after passing through a small induction fan. The fan is sized to provide sufficient air flow only when the vehicles ram air intake is reduced.

A small dc motor with a double-output through-shaft drives the fan on one end, and the boost pump and gerotor turbine lubrication pump on the other end. The assembly, as shown in Figure 20, also provides for an air-scoop device that could be used to supply condenser heated air $(120^{\circ}F)$ to the cab interior for passenger comfort heating and for window defrosting. The overall length of the integrated condenser-regenerator unit is approximately the same as the vapor-generator (4 feet) and is installed in a similar manner - to the rear of the vehicle on the opposite side to the vapor-generator.

Figure 21 shows the RC-1 bottoming cycle system installed on a heavy-duty vehicle.



Figure 19. Power Conversion Unit (PCU)





Figure 20. Condenser-Regenerator Module





A parts list for the RC-1 system is presented in Table 16, listing the major assemblies and part breakdown for each assembly. This table also lists the estimated weight of the various components of the system. The total weight (692 lb) given here does not consider the plumbing, control system, and RC-1 organic fluid inventory. Table 17 organizes the RC-1 Bottoming Cycle System weight into the three (3) basic modules and also includes the plumbing, control system, and RC-1 fluid inventory in the estimate. The net ORCS weight of 740 lb is 300 lb lighter than the 1039-lb Fluorinol Truck Bottoming Cycle System.

2.4 LIFE-CYCLE COST ANALYSIS

After selection of the design point establishing the system performance and definition of the components by the system design, a cost analysis was conducted for this RC-1 Organic Rankine Bottoming Cycle System.

Efforts in the cost evaluation for the RC-1 Bottoming Cycle included:

- Potential Capital Cost
- Maintenance Cost
- Simple Payback Period
- Return on Investment

The following sections describe the evaluations and estimates made, including methodology and the economic and operational reference data provided by NASA Lewis in arriving at the final results.

2.4.1 Bottoming Cycle Manufacturing Cost

In 1980 Thermo Electron Corporation contracted Rath & Strong, Inc., under PO #42646, to develop the manufacturing costs to produce a dieselorganic Rankine compound engine for long-haul trucks. At that time the manufacturing costs were developed for three (3) different annual production rates and are shown in the following table.

Annual Production Rate (units/yr)	Total Manufacturing (\$/unit)
5,000	· 5619
150,000	1934

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TABLE 16

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PARTS LIST FOR RC-1 ORGANIC RANKINE-CYCLE SYSTEM

				Est.	Weight	
Item	Description	Drawing No.	Req.	(1	5)	Remarks
				Part	Ass'y.	
1	Turbine Ass'y. Exhaust Housing Inlet/Seal Housing Rotor Shaft Bearings Seal Ass'y. Preload Spacer Mounting Flange	048-01-000	1	5.69 19.30 0.55 0.48 0.27 0.12 1.47 5.65	33.53	
2	Turbine Gearbox Housing Gears Coupling Quill-Shaft Bearings	048-02-000	· 1		36	Actual weight of similar gearbox
3	Feedpump Drive Housing Over-Running Clutch Flex-Shaft Bearings Pulleys Belt	048-03-000	1	- 25 -9	36	Housing weight included in turbine gearbox

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TABLE 16 (Cont'd)

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PARTS LIST FOR RC-1 ORGANIC RANKINE-CYCLE SYSTEM

				Est. Weight (lb)		•
ltem	Description	Drawing No.	Req.			Remarks
				Part	Ass'y.	
4	Feedpump Ass'y. Housing Roll-Nut Pistons Bearings Potentiometer Motor	048-04-000	1		17.3	Actual weight of F1-85 feedpump
5	Vapor Generator Shell Baseplate Top Exhaust Pipe Core (Finned Tube) Diverter Valve	048-05-000	1		450	Actual weight
6	Cond./Regen. Ass'y. Shell Top Plate Regen. Ass'y. Tubes Core (Finned Tube)	048-06-000	1	22 13 40	75	
7	Fan Ass'y.	048-07-000	1	*	1.5	

A-8530c

TABLE 16 (Cont'd)

PARTS LIST FOR RC-1 ORGANIC RANKINE-CYCLE SYSTEM

ltem	Description	Drawing No.	Req.	Est. Weight (lb)		Remarks
				Part	Ass'y.	
8	<u>Filter Ass'y.</u> Shell Flanges Element	048-08-000	1.		5	Actual weight
9	Boost/Lube Pump Motor Housing Shaft Empellar Bearings Gerotor Seal Ass'y.	048-09-000	1		23.8	Actual weight
10	<u>Oil Cooler</u> Frame Core (Finned Tube)	048-10-000	1		2.5	
11	Regen. Support Brkt.	048-00-011	1		9.5	
12	Filter/Boost Pump Brkt.	048-00-11	1		2.0	
					692.13	Total Weight

TABLE 17

RC-1 BOTTOMING CYCLE SYSTEM WEIGHT

•	Vapor-Generator Module	450 lb		
۰ ی	Power Conversion Unit (PCU)	135 lb		
	- Turbine			
	- Gearbox			
	- Feedpump			
	- Feedpump Drive			
	- Plumbing and Hardware			
٠	Condenser-Regenerator Module	119 lb		
	~ Condenser			
	~ Regenerator			
	~ Fan			
	- Filter Assembly			
	- Boost/Lube Pump			
	~ Oil Cooler			
	 Regenerator Support Bracket 			
	- Filter/Boost Pump Bracket			
٠	Control System	7 Ib		
RC-1 Fluid Inventory				
	(4 gai at 13.4 lb/gai)	<u>54 lb</u>		
	Gross ORCS Weight	765 lb		
	Deduct for Muffler	<u>-25 lb</u>		
	Net ORCS Weight	740 lb		

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The factors and concepts applied in the calculation of the above unit costs are summarized as follows.

• Material Cost

The material cost per unit applies to the 5000 per year volume level with a 0.9 learning curve concept applied for the other production levels. That is, for every doubled quantity, the cost is reduced by 10 percent.

• Direct Labor Cost

A labor rate of \$8.00 per hour was used for each man-hour.

• Indirect Labor Cost

40 percent of direct labor unit cost was used.

• Tools Cost

These costs were amortized over 3 years.

• Equipment Cost

Machine Utilization - For the 5000 units per year volume level, it was estimated that the machines would be 30 percent loaded on an ideal basis. Industry experience indicates that only about 75 percent of ideal utilization is attained; therefore, a combined factor of 40 percent (30% ÷ 75%) has been applied. A multiplication factor of 0.90 was used for 50,000 units/year and 0.95 was used for 150,000 units/year.

- The resultant estimates were amortized over 12 years.

A manufacturing cost of \$4189 for the RC-1 Bottoming Cycle System has been developed for an annual production rate of 10,000 units (Table 18). For all similar components in the RC-1 system design, the same methodology was used as described above for the Rath & Strong, Inc. cost development. New cost estimates or vendor quoted costs (noted items in Table 18) for components that are unique or different for the RC-1 system design have been incorporated. Factors for the 10,000 units/year production level have been calculated or extrapolated from those used in the Rath & Strong cost estimates. Those costs shown in Table 18 have also been escalated by 20.4 percent to bring the estimates in line with the current dollar value. This 20.4-percent factor was obtained from the "Survey of Current Business," United States Department of Commerce/ Bureau of Economic Analysis and is the inflation index for Transportation Motor Vehicles and Equipment.

MANUFACTURING COST ESTIMATE FOR RC-1 BOTTOMING CYCLE SYSTEM AT AN ANNUAL PRODUCTION RATE OF 10,000 UNITS

Assembly	Dollars Per Unit
Feedpump	\$ 402
Turbine/Gearbox	827
Condenser/Regenerator	490
Vapor Generator	1500*
Feedpump Drive	102
Boost/Lube Pump and Fan	205
Filter Assembly	36
Oil Cooler	11
Rupture Disc Assembly	85
Bracket Support Regenerator	17
Flanges, Condenser	6
Miscellaneous Equipment and Hardware - Corrugated Metal Hose - "O"-Rings - Tubing - Pipe - Hardware	27
Control System	305
RC-1 Organic Working Fluid	<u> 176</u> **
Total Cost	\$4189

Notes:

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*This cost is an actual quote from Cannon Boiler Works of Pennsylvania. The recalculated cost for this heat exchanger performed by the Rath & Strong method yielded a cost of \$1405 for this production level. This adds another degree of confidence to the Rath & Strong method.

**The fluid inventory required for the RC-1 bottoming cycle system is 4 gallons at an estimated manufacturing cost of \$44 per gallon. The cost estimate of \$44 per gallon for the RC-1 fluid is based on an annual production rate of 100,000 gallons. This RC-1 manufacturing cost has been developed utilizing both in-house experience and a Monsanto study (ref. 3). One input for this cost estimate was the experience of an actual pilot plant built and operated by Thermo Electron Corporation with production rate of 66 gallons per year. The total manufacturing cost at this small production capacity was \$1208 per gallon. Estimates of the manufacturing cost for a plant with a production capacity 100 times greater (6600 gallons per year) resulted in a cost at \$86 per gallon. Recent verbal communication with Imperial Smelting of England indicates that RC-1 fluid can be purchased from them at approximately \$1000 per gallon in single gallon quantities.

The other major source and basis for the RC-1 manufacturing cost developed for this system is the previously mentioned Monsanto study. A manufacturing and economic evaluation indicated a plant selling price of 1.40/lb of RC-1 for a large plant producing 50,000.000 lb/hr. With a liquid density of 13 lb/gal, these values are equivalent to \$18/gal RC-1 and 3,900,000 gal/year.

2.4.2 Bottoming Cycle Maintenance Cost

A maintenance expense for the RC-1 Bottoming Cycle System has been estimated at \$0.011 per mile. This maintenance cost has been derived from prior evaluation and studies performed for the Fluorinol Truck Bottoming Cycle System (\$0.01/mile in 1980 dollars), and the actual service contract price for maintenance that Thermo Electron offers for its TECOGENTM Cogeneration Module.

The TECOGEN is an automatic cogeneration module manufactured by Thermo Electron. The module simultaneously produces hot water and 60 kW of electricity. This powerplant, although a stationary system, can be compared to the bottoming cycle with similar components. The cogeneration module, which is designed to operate 24 hours a day, continuously, contains the following major components.

- Modified V-8 Engine for Industrial Use (Not in Bottoming Cycle System)
- Engine Coolant Heat Exchanger (Coolant to Water)
- Exhaust Heat Exchanger (Gas to Water)
- Water-Cooled Exhaust Manifolds
- Oil Cooler
- Induction Generator (Not in Bottoming Cycle System)

- Microprocessor Control System
 - Automatically starts and stops the unit
 - , Connects or disconnects the module with the electric utility
 - Provides diagnostic capability

Thermo Electron offers a service maintenance contract at a cost of \$0.015/kWh. The maintenance price includes:

- Personnel travel to and from the site.
- Preventive maintenance schedule that includes engine oil changes every 500 hours.
- Repairs and parts replacement, both scheduled and unscheduled.
- Complete engine replacement after 12,000 hours (if required).

Personnel travel to and from the site is one-third the maintenance cost. Because this cost does not apply for the bottoming cycle where maintenance is performed in a service area, then the comparable maintenance cost for the bottoming cycle can be based on two-thirds the module maintenance cost $(2/3 \times \$0.015/kWh)$, or \$0.010/kWh. For continuous operation of the module the total maintenance cost for the year then becomes:

 $0.01/kWh \ge 60 kW \ge 8760 hr/yr = $5256/yr$.

For steady-state operation this translates to an equivalent yearly mileage of:

8760 hr/yr x 55 miles/hr = 481,800 miles/yr.

Thus, the maintenance cost calculated on a mileage basis becomes:

 $\frac{\$5256/\text{yr}}{481,800 \text{ miles/yr}} = \$0.0109/\text{mile}.$

The 1980 \$0.01/mile maintenance cost for the Truck Bottoming Cycle System considered the following factors.

- Scheduled Maintenance
 - Vapor-generator cleaning (water wash) at same interval as diesel engine oil change (Mack recommends oil change every 16,000 miles).
 - Gearbox oil changes at same interval as diesel engine oil change.
 - Organic fluid filter cartridge change at same interval as diesel engine oil change.

- One (1) burst disc replacement per year
- One-half organic fluid inventory replacement per year (2 to $2\frac{1}{2}$ gallons)
- Unscheduled Maintenance
 - One (1) fan belt replacement
 - Control system repair (i.e., card replacement).
 - Other failures (i.e., organic fluid flexline leak, turbine seal leak, and feedpump feedback pot)

Time estimates for the normal maintenance schedule were determined from actual operating experience with the Test Bed Vehicles (over 55,000 miles) and also with the laboratory system (over 2000 hours of operation). These times together, with allowance for some unscheduled maintenance, were used to determine the prior 1980 maintenance cost of \$0.01/mile. Of this \$0.01/mile maintenance burden for the ORCS, just over half (\$0.006/mile) is for the normal maintenance schedule, while the balance of the \$0.01/mile cost is allocated for unscheduled maintenance or failures. These maintenance costs are based on a truck annual mileage of 100,000 miles.

Considering the above mentioned service contract for the TECOGEN and prior studies done on the Truck Bottoming Cycle System to determine maintenance requirements both scheduled and unscheduled, a maintenance cost of \$0.011 per mile was established for this study.

2.4.3 Bottoming Cycle System Economic Incentives

The economic benefits of an RC-1 Bottoming Cycle System are based on the fuel savings potential of the system, system cost, and economic conditions that exist at the time. The fuel savings potential has been established by the analysis task of this program and the unit and other associated costs have been developed in the preceding paragraphs. The economic conditions for this study have been established by the NASA Lewis Research Center. Table 19 shows the NASA Reference Diesel Engine Data with respect to the mission fuel economy and selling price for each of the four (4) base diesel engine configurations. The other pertinent economic and operational data, such as fuel price and annual vehicle mileage, are presented in Table 20.

The simple payback period was then calculated for the Bottoming Cycle versus the Turbocompound Engine System by the procedure designated by NASA and shown in Table 21. This example calculates the payback period of the bottoming cycle compound engine only in reference to

TABLE 19NASA REFERENCE DIESEL ENGINE DATA

DIESEL CONFIGURATION	BHP*	BSFC (LB/BHP-HR)	MISSION (MPG)	SELLING PRICE (\$)
тс	317	0.315	5.66	14,000
TC/A	320	0.310	5.75	14,500
TCPD	335	0.297	6.00	16,000
TCPD/A	340	0.293	6.08	16,500

*FOR ENGINE RATINGS ABOVE THOSE LISTED IN THE TABLE, ADD \$30 PER HORSEPOWER.

TABLE 20

NASA REFERENCE ECONOMIC/OPERATIONAL DATA

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•	COST OF MONEY (INTEREST)	128
•	CORPORATE TAX RATE	468
•	INVESTMENT TAX CREDIT	78
•	ESCALATION RATE	08
•	DIESEL FUEL PRICE PER GALLON	\$1.20
•	ANNUAL MILEAGE PER TRUCK	100,000
•	HARDWARE USEFUL LIFE	7 YEARS
•	ANNUAL PRODUCTION RATE	10,000*
٠	SELLING PRICE/MFG COST	2.0

*10% PENETRATION ON A MARKET OF 100,000 CLASS-8 TRUCKS SOLD PER YEAR

TABLE 21

NASA REFERENCE SIMPLE PAYBACK PERIOD (BOTTOMING CYCLE VS TURBOCOMPOUND)

- BOTTOMING CYCLE PARAMETERS:
 - OUTPUT AS APPLIED TO TC ENGINE 56 HP
 - BOTTOMING CYCLE SELLING PRICE \$8378

• COMPOUND ENGINE PARAMETERS:

- RATED BHP 317 + 56 = 373
- BSFC/MPG 0.268/6.65
- ANNUAL FUEL 15,037 GAL/\$18,044
- ENGINE SYSTEM PRICE \$14,000 + \$8378 = \$22,378
- COMPARABLE TCPD/A ENGINE PARAMETERS:
 - RATED BHP 373
 - BSFC/MPG 0.293/6.08
 - ANNUAL FUEL 16,447 GAL/\$19,737
 - ENGINE SYSTEM PRICE $$16,500 + (33 \times 30) = $17,490$

• SIMPLE PAYBACK :

- ANNUAL FUEL SAVINGS 1410 GAL/\$1692
- ENGINE CAPITAL COST DIFFERENCE \$4888
- PAYBACK PERIOD 2.89 YEARS

the turbocompound (TCPD/A) engine and does not consider the payback period requirements of the TCPD/A engine with reference to the TC engine configuration, for example. A set of calculations for payback period was made for both the TCPD/A engine and the bottoming cycle compound engine versus each reference diesel engine configuration. The examples for each case are presented in Tables 22 (a) through 22(d). The last item in each table shows the difference in payback period between the Bottoming Cycle Compound Engine and the TCPD/A Engine Systems.

A summary for the simple payback period for the option choice versus each base engine configuration is shown in Table 23. Besides the payback periods for the TCPD/A, ORCS, and the difference between them, the payback periods for the other engine options versus the reference base engine configuration has also been calculated where applicable. If a potential buyer is weighing the options against the base TC engine, consideration should be given to the payback period of each option for the fuel economy benefits derived and the capital cost difference required. For instance, if one chooses the TC/A engine over the TC engine, the simple payback period is 1.22 years for the fuel saving benefit of the TC/A over the TC engine for the \$500 cost difference between them. Similarly, if one decides to spend \$8400 for the ORCS option, it will take only 2.12 years to pay back considering the fuel savings potential of this system.

A nomogram was developed to graphically determine the simple payback period of any Bottoming Cycle System compared to any of the four (4) reference diesel engine configurations knowing only the compound engine BSFC and the additional power output provided by the bottoming system. The nomogram and expressions for the development of this payback period nomogram are presented in Appendix D.

The simple payback period examples shown above were based on a fuel price of \$1.20/gal and an annual truck mileage of 100,000 miles (defined by NASA reference data). Figure 22 shows the payback period sensitivity to fuel price for the vehicle mileage "window" from 100,000 to 150,000 miles per year. As vehicle annual mileage approaches 150,000 miles and fuel prices advance towards the \$2.00 per gallon figure, this \$8400 RC-1 Bottoming Cycle System will pay back in less than one (1) year.

A cash flow projection was executed for the RC-1 Bottoming Cycle System utilizing the Table 20 economic reference data furnished by NASA. Table 24 shows the cash flow subset of a potential fleet or other vehicle owner for the 7-year hardware useful life with cumulative profits of over \$6600 and cash flow exceeding \$5500. This example shows an internal rate of return of 45 percent for this "up front" capital investment, even after introducing a maintenance cost of \$0.011 per mile.
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TABLE 22a

SIMPLE PAYBACK EXAMPLE FOR BOTTOMING TC ENGINE

) BOTTOMING CYCLE PARAMETERS

-	Output as	Applied to TC Engine	-	56 hp
-	Bottoming	Cycle Selling Price	-	\$8378

2) COMPOUND ENGINE PARAMETERS (BOTTOMING CYCLE)

-	Rated bhp	-	317 + 56 = 373 hp
-	bsfc/mpg	-	0.268/6.65
-	Annual Fuel	-	15,037 gal/\$18,044
-	Engine System Price	-	\$14.000 + \$8.378 = \$22.378

3) SIMPLE PAYBACK (BOTTOMING CYCLE)

- Annual Fuel Savings	-	2631 gal/\$3157
- Engine Capital Cost Difference	-	\$6698
- Payback Period	-	2.12 years

4) DIESEL ENGINE PARAMETERS

Output of TC Engine
bsfc/mpg
Annual Fuel
Engine Selling Price
317 + 56 = 373 hp
0.315/5.66
17,668 gal/\$21,201
\$14,000 + (56 x 30) = \$15,680

5) TCPD/A ENGINE PARAMETERS

Rated bhp
bsfc/mpg
Annual Fuel
Engine System Price
- 340 + 33 = 373 hp
0.293/6.08
16,447 gal/\$19,737
\$16,500 + (33 x 30) = \$17,490

6) SIMPLE PAYBACK (TURBOCOMPOUND)

- Annual Fuel Savings	-	1221 gal/\$1465
- Engine Capital Cost Difference	. –	\$1810
– Payback Period	-	1.24 years

7) DIFFERENCE IN PAYBACK PERIOD

(3) (6)2.12 - 1.24 = 0.88 year

TABLE 22b

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SIMPLE PAYBACK EXAMPLE FOR BOTTOMING TC/A ENGINE

1) BOTTOMING CYCLE PARAMETERS		
- Output as Applied to TC/A Engine	-	52
- Bottoming Cycle Selling Price	-	\$8378
2 COMPOUND ENGINE PARAMETERS	(BC	OTTOMING CYCLE)
- Rated bhp - bsfc/mpg - Annual Fuel - Engine System Price		320 + 52 = 372 hp 0.267/6.68 \$14,970 gal/\$17,965 \$14,500 + \$8,378 = \$22,878
3 SIMPLE PAYBACK (BOTTOMING C	YCL	JE)
 Annual Fuel Savings Engine Capital Cost Difference Payback Period 	-	2421 gal/\$2905 \$6818 2.35 years
4 DIESEL ENGINE PARAMETERS		
 Output of TC/A Engine bsfc/mpg Annual Fuel Engine Selling Price 	- - -	320 + 52 = 372 hp 0.310/5.75 17,391 gal/\$20,870 \$14,500 + (52 x 30) = \$16,060
5 TCPD/A ENGINE PARAMETERS		
- Rated bhp - bsfc/mpg - Annual Fuel - Engine System Price	- - -	340 + 32 = 372 hp 0.293/6.08 16,447 gal/\$19,737 \$16,500 + (32 x 30) = \$17,460
6 SIMPLE PAYBACK (TURBOCOMPO	UND))
 Annual Fuel Savings Engine Capital Cost Difference Payback Period 	- - -	944 gal/\$1133 \$1400 1.24 years
7 DIFFERENCE IN PAYBACK PERIOT 3 6 2.35 - 1.24 = 1.11 years)	
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TABLE 22c

SIMPLE PAYBACK EXAMPLE FOR BOTTOMING TCPD ENGINE

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 Output as Applied to TCPD - 50 hp Engine Bottoming Cycle Selling Price - \$8378 COMPOUND ENGINE PARAMETERS (BOTTOMING CYCLE) Rated bhp - 335 + 50 = 383 hp bsfc/mpg - 0.258/6.91 Annual Fuel - 14,472 gal/\$17,366 Engine System Price - \$16,000 + \$8,378 = \$24,378 SIMPLE PAYBACK (BOTTOMING CYCLE) Annual Fuel Savings - 2195 gal/\$2634 Engine Capital Cost Difference - \$6878 Payback Period - 2.61 years DIESEL ENGINE PARAMETERS Output of TCPD Engine - 335 + 50 = 385 hp bsfc/mpg - 0.297/6.00 Annual Fuel - 16,667 gal/\$20,000 Engine Selling Price - \$16,000 + (50 x 30) = \$17, TCPD/A ENGINE PARAMETERS Rated bhp - 340 + 45 = 385 hp bsfc/mpg - 0.293/6.08 Annual Fuel - 16,447 gal/\$19,737 Engine System Price - \$16,500 + (45 x 30) = \$17, SIMPLE PAYBACK (TURBOCOMPOUND) Annual Fuel Savings - 220 gal/\$264 Engine Capital Cost Difference - \$350 Payback Period - 1.33 years DIFFERENCE IN PAYBACK PERIOD (3) (6) 2.61 - 1.33 = 1.28 years 	(1)	BOTTOMING CYCLE PARAMETERS		
Bottoming Cycle Selling Price - \$8378 2 COMPOUND ENGINE PARAMETERS (BOTTOMING CYCLE) - Rated bhp - 335 + 50 = 383 hp - bsfc/mpg - 0.258/6.91 - Annual Fuel - 14,472 gal/\$17,366 - Engine System Price - \$16,000 + \$8,378 = \$24,378 3 SIMPLE PAYBACK (BOTTOMING CYCLE) - Annual Fuel Savings - 2195 gal/\$2634 - Engine Capital Cost Difference - \$6678 - Payback Period - 2.61 years 4 DIESEL ENGINE PARAMETERS - Output of TCPD Engine - 335 + 50 = 385 hp - bsfc/mpg - 0.297/6.00 - Annual Fuel - 16,667 gal/\$20,000 - Engine Selling Price - \$16,000 + (50 x 30) = \$17, 5 TCPD/A ENGINE PARAMETERS - Rated bhp - 340 + 45 = 385 hp - bsfc/mpg - 0.293/6.08 - Annual Fuel - 16,447 gal/\$19,737 - Engine System Price - \$16,500 + (45 x 30) = \$17, 6 SIMPLE PAYBACK (TURBOCOMPOUND) - Annual Fuel Savings - 220 gal/\$264 - Engine Capital Cost Difference - \$350 - Payback Period - 1.33 years 7 DIFFERENCE IN PAYBACK PERIOD 3 (6) 2.61 - 1.33 = 1.28 years	-	- Output as Applied to TCPD	-	50 hp
 COMPOUND ENGINE PARAMETERS (BOTTOMING CYCLE) Rated bhp bsfc/mpg 0.258/6.91 Annual Fuel 14,472 gal/\$17,366 Engine System Price \$16,000 + \$8,378 = \$24,378 SIMPLE PAYBACK (BOTTOMING CYCLE) Annual Fuel Savings 2195 gal/\$2634 Engine Capital Cost Difference \$6878 Payback Period 2.61 years DIESEL ENGINE PARAMETERS Output of TCPD Engine 325 + 50 = 385 hp bsfc/mpg 0.297/6.00 Annual Fuel 16,667 gal/\$20,000 Engine Selling Price \$16,000 + (50 x 30) = \$17, TCPD /A ENGINE PARAMETERS Rated bhp 0.293/6.08 Annual Fuel 16,447 gal/\$19,737 Engine System Price \$16,500 + (45 x 30) = \$17, SIMPLE PAYBACK (TURBOCOMPOUND) Annual Fuel 16,447 gal/\$19,737 Engine System Price \$16,500 + (45 x 30) = \$17, SIMPLE PAYBACK (TURBOCOMPOUND) Annual Fuel Savings 220 gal/\$264 Engine Capital Cost Difference \$350 Payback Period 1.33 years DIFFERENCE IN PAYBACK PERIOD (3) (6) 2.61 - 1.33 = 1.28 years 		- Bottoming Cycle Selling Price	-	\$8378
- Rated bhp - bsfc/mpg - Annual Fuel - Annual Fuel - Engine System Price - 14,472 gal/\$17,366 - Engine System Price - 14,472 gal/\$17,366 - Engine System Price - 14,472 gal/\$17,366 - Engine System Price - 14,472 gal/\$2634 - Engine Savings - 2195 gal/\$2634 - Engine Capital Cost Difference - \$6878 - Payback Period - 2.61 years (1) DIESEL ENGINE PARAMETERS - Output of TCPD Engine - bsfc/mpg - 0.297/6.00 - Annual Fuel - 16,667 gal/\$20,000 - Engine Selling Price - \$16,000 + (50 x 30) = \$17, (2) TCPD/A ENGINE PARAMETERS - Rated bhp - bsfc/mpg - 0.293/6.08 - Annual Fuel - 16,447 gal/\$19,737 - Engine System Price - \$16,500 + (45 x 30) = \$17, (4) SIMPLE PAYBACK (TURBOCOMPOUND) - Annual Fuel Savings - Payback Period - Annual Fuel Savings - Payback Period - Annual Fuel Savings - Payback Period - Annual Fuel Savings - 220 gal/\$264 - Engine Capital Cost Difference - \$350 - Payback Period - 1.33 years (1) DIFFERENCE IN PAYBACK PERIOD (3) (6) 2.61 - 1.33 = 1.28 years	2	COMPOUND ENGINE PARAMETERS	(BC	TTOMING CYCLE)
 3 SIMPLE PAYBACK (BOTTOMING CYCLE) Annual Fuel Savings Engine Capital Cost Difference \$6878 Payback Period 2.61 years 4 DIESEL ENGINE PARAMETERS Output of TCPD Engine astronomy of the state of the s		 Rated bhp bsfc/mpg Annual Fuel Engine System Price 	- - -	335 + 50 = 383 hp 0.258/6.91 14,472 gal/\$17,366 \$16,000 + \$8,378 = \$24,378
- Annual Fuel Savings - Engine Capital Cost Difference - Payback Period - Output of TCPD Engine - bsfc/mpg - Coutput of TCPD Engine - bsfc/mpg - Annual Fuel - Engine Selling Price - Rated bhp - bsfc/mpg - Coutput of TCPD Engine - 335 + 50 = 385 hp - 0.297/6.00 - 16,667 gal/\$20,000 - Engine Selling Price - 340 + 45 = 385 hp - 0.293/6.08 - Annual Fuel - 340 + 45 = 385 hp - 0.293/6.08 - Annual Fuel - 16,447 gal/\$19,737 - Engine System Price - \$16,500 + (45 x 30) = \$17, - Annual Fuel Savings - 220 gal/\$264 - Engine Capital Cost Difference - \$350 - Payback Period - 1.33 years - DIFFERENCE IN PAYBACK PERIOD - 33 6 2.61 - 1.33 = 1.28 years	3	SIMPLE PAYBACK (BOTTOMING C	YCL	E)
 (4) DIESEL ENGINE PARAMETERS Output of TCPD Engine bsfc/mpg 0.297/6.00 Annual Fuel 16,667 gal/\$20,000 Engine Selling Price \$16,000 + (50 x 30) = \$17, (5) TCPD/A ENGINE PARAMETERS Rated bhp 340 + 45 = 385 hp bsfc/mpg 0.293/6.08 Annual Fuel 16,447 gal/\$19,737 Engine System Price \$16,500 + (45 x 30) = \$17, (6) SIMPLE PAYBACK (TURBOCOMPOUND) Annual Fuel Savings 220 gal/\$264 Engine Capital Cost Difference \$350 Payback Period 1.33 years (7) DIFFERENCE IN PAYBACK PERIOD (3) (6) 2.61 - 1.33 = 1.28 years 		 Annual Fuel Savings Engine Capital Cost Difference Payback Period 	- - -	2195 gal/\$2634 \$6878 2.61 years
 Output of TCPD Engine - 335 + 50 = 385 hp bsfc/mpg - 0.297/6.00 Annual Fuel - 16,667 gal/\$20,000 Engine Selling Price - \$16,000 + (50 x 30) = \$17, TCPD /A ENGINE PARAMETERS Rated bhp - 340 + 45 = 385 hp bsfc/mpg - 0.293/6.08 Annual Fuel - 16,447 gal/\$19,737 Engine System Price - \$16,500 + (45 x 30) = \$17, SIMPLE PAYBACK (TURBOCOMPOUND) Annual Fuel Savings - 220 gal/\$264 Engine Capital Cost Difference - \$350 Payback Period - 1.33 years DIFFERENCE IN PAYBACK PERIOD (3) (6) 	4	DIESEL ENGINE PARAMETERS		
 TCPD /A ENGINE PARAMETERS Rated bhp bsfc/mpg 0.293/6.08 Annual Fuel 16,447 gal/\$19,737 Engine System Price \$16,500 + (45 x 30) = \$17, SIMPLE PAYBACK (TURBOCOMPOUND) Annual Fuel Savings 220 gal/\$264 Engine Capital Cost Difference \$350 Payback Period 1.33 years DIFFERENCE IN PAYBACK PERIOD 	-	 Output of TCPD Engine bsfc/mpg Annual Fuel Engine Selling Price 	- - -	335 + 50 = 385 hp 0.297/6.00 16,667 gal/\$20,000 \$16,000 + (50 x 30) = \$17,500
 Rated bhp bsfc/mpg Annual Fuel Engine System Price SIMPLE PAYBACK (TURBOCOMPOUND) Annual Fuel Savings Engine Capital Cost Difference Fayback Period 1.33 years DIFFERENCE IN PAYBACK PERIOD (3) (6) 	5	TCPD/A ENGINE PARAMETERS		
 6 SIMPLE PAYBACK (TURBOCOMPOUND) Annual Fuel Savings Engine Capital Cost Difference Payback Period 1.33 years 7 DIFFERENCE IN PAYBACK PERIOD 3 6 2.61 - 1.33 = 1.28 years 		 Rated bhp bsfc/mpg Annual Fuel Engine System Price 	- - -	340 + 45 = 385 hp 0.293/6.08 16,447 gal/\$19,737 \$16,500 + (45 x 30) = \$17,850
 Annual Fuel Savings - 220 gal/\$264 Engine Capital Cost Difference - \$350 Payback Period - 1.33 years 7 DIFFERENCE IN PAYBACK PERIOD 3 6 2.61 - 1.33 = 1.28 years 	6	SIMPLE PAYBACK (TURBOCOMPOU	JND)
7 DIFFERENCE IN PAYBACK PERIOD 3 6 2.61 - 1.33 = 1.28 years	-	 Annual Fuel Savings Engine Capital Cost Difference Payback Period 		220 gal/\$264 \$350 1.33 years
	7	DIFFERENCE IN PAYBACK PERIOD 3 $62.61 - 1.33 = 1.28$ years)	

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TABLE 22d

SIMPLE PAYBACK EXAMPLE FOR BOTTOMING TCPD/A ENGINE

$\left(1\right)$	BOTTOMING CYCLE PARAMETERS		
-	- Output as Applied to TCPD/A	-	46 hp
	- Bottoming Cycle Selling Price	-	\$8378
(2)	COMPOUND ENGINE PARAMETERS	(BC	TTOMING CYCLE)
-	 Rated bhp bsfc/mpg Annual Fuel Engine System Price 		340 + 46 = 386 hp 0.258/6.91 14,472 gal/\$17,366 \$16,500 + \$8,378 = \$24,878
3	SIMPLE PAYBACK (BOTTOMING C	YCL	E)
	– Annual Fuel Savings – Engine Capital Cost Difference – Payback Period	- - -	1975 gal/\$2370 \$6998 2.95 years
(4)	DIESEL ENGINE PARAMETERS		
0	 Output of TCPD/A Engine bsfc/mpg Annual Fuel Engine Selling Price 	- - -	340 + 46 = 386 hp 0.293/6.08 16,447 gal/\$19,737 \$16,500 + (46 x 30) = \$17,880
(5)	TCPD/A ENGINE PARAMETERS		
	- Rated bhp - bsfc/mpg - Annual Fuel - Engine System Price		340 + 46 = 386 hp 0.293/6.08 16,447 gal/\$19,737 \$16,500 + (46 x 30) = \$17,880
(6)	SIMPLE PAYBACK (TURBOCOMPOU	JND)
)	 Annual Fuel Savings Engine Capital Cost Difference Payback Period 		

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(7) DIFFERENCE IN PAYBACK PERIOD

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TABLE 23 SIMPLE PAYBACK PERIOD FOR OPTION CHOICE (YEARS)

BASE ENGINE			OPTION			ΔPAYBACK PERIOD
CONFIGURATION	тс	TC/A	TCPD	TCPD/A	ORCS	ORCS-TCPD/A
тс	-	1.22	1.20	1.24	2.12	0.88
TC/A	-	· -	1.20	1.24	2.35	1.11
TCPD	- -	-	-	1.33	2.61	1.28
TCPD/A	_	-	_	-	2.95	2.95

309-1083



Figure 22. RC-1 Bottoming Cycle System Simple Payback Period without Maintenance Burden

TABLE 24

RC-1 BOTTOMING CYCLE SYSTEM - CASH FLOW PROJECTION

COST OF MONEY	128
CORPORATE TAX RATE	468
INVESTMENT TAX CREDIT	78
ESCALATION RATE	08
DIESEL FUEL PRICE/GAL	\$1.20
ANNUAL MILEAGE/TRUCK	100,000
HARDWARE USEFUL LIFE (YR)	7
OPERATION AND MAINTENANCE COST/MIL	\$0.011
ORCS COST	\$8378.00
ORCS COST DIFFERENTIAL	\$6698.00
ORCS POWER OUTPUT (HP)	56
COST/HP ABOVE BASE ENGINE POWER	\$30.00
MPG BASE DIESEL	5.66
MPG ORCS/DIESEL	6.65
DEPRECIATION RATE/YR	38
DEBT	08
LOAN TERM (MON)	0

YEAR	1	2	3	4	5	6	7
ORCS COST ORCS COST DIFFERENTIAL INVESTMENT TAX CREDIT DEPRECIATION FUEL SAVINGS OPERATION AND MAINTENANCE COST OPERATING SAVINGS LOAN PRINCIPAL/END MONTHS ON NOTE PRINCIPAL PAYMENT INTEREST PAYMENT TOTAL: PRINCIPAL AND INTEREST	8378 6698 586 2793 3156 1100 2056 0 0 0 0 0 0 0 0 0 0	0 0 2793 3156 1100 2056 0 0 0 0 0	0 0 2793 3156 1100 2056 0 0 0 0 0	0 0 3156 1100 2056 0 0 0 0 0	0 0 3156 1100 2056 0 0 0 0	0 0 3156 1100 2056 0 0 0 0	0 0 3156 1100 2056 0 0 0 0
PRETAX PROFITS TAXES AFTERTAX PROFITS CASH FLOW CUMULATIVE PROFITS CUMULATIVE CASH FLOW	- 7 36 - 925 189 - 3717 189 - 3717	- 7 36 - 339 - 398 2395 - 209 - 1 321	-736 -339 -398 2395 2186 1074	2056 946 1110 1110 3297 2184	2056 946 1110 1110 4407 3294	2056 946 1110 1110 5517 4405	2056 946 1110 1110 6628 5515

3.26

Similar cash flow projections are presented in Tables 25 and 26. A 50-percent loan for 5 years (60 months) is carried in the Table 25 example, while the full capital cost (100 percent debt) is carried for 60 months in the example of Table 26. The internal rate of return for the 50-percent debt of Table 25 exceeds 200 percent while the 100-percent debt cash flow projection of Table 26 indicates a rate of return that approaches infinity. The reason for this high rate of return in Table 26 is that the cash flow is never negative. By definition the rate of return equates the equivalent value of positive cash flow to the negative cash flow over the number of years (N) in the study period and determines the interest rate value (i).

TABLE 25

RC-1 BOTTOMING CYCLE SYSTEM - CASH FLOW PROJECTION

.

COST OF MONEY	128
CORPORATE TAX RATE	468
INVESTMENT TAX CREDIT	78
ESCALATION RATE	08
DIESEL FUEL PRICE/GAL	\$1.20
ANNUAL MILEAGE/TRUCK	100,000
HARDWARE USEFUL LIFE (YR)	7
OPERATION AND MAINTENANCE COST/MIL	\$0.011
ORCS COST	\$8378.00
ORCS COST DIFFERENTIAL	\$6698.00
ORCS POWER OUTPUT (HP)	56
COST/HP ABOVE BASE ENGINE POWER	\$30.00
MPG BASE DIESEL	5.66
MPG ORCS/DIESEL	6.65
DEPRECIATION RATE/YR	38
DEBT	50%
LOAN TERM (MON)	60

YEAR	1	2	3	4	5	6	7
ORCS COST	4189	0	0	0	0	0	0
ORCS COST DIFFERENTIAL	3349	0	0	0	0	0	0.
INVESTMENT TAX CREDIT	586	0	0	0	0	0	0
DEPRECIATION	2793	2793	2793	0	0	0	0
FUEL SAVINGS	3156	3156	3156	3156	3156	3156	3156
OPERATION AND MAINTENANCE COST	1100	1100	1100	1100	1100	1100	1100
OPERATING SAVINGS	2056	2056	2056	2056	2056	2056	2056
LOAN PRINCIPAL/END	2829	2243	1583	838	0	0	0
MONTHS ON NOTE	48	36	24	12	0	0	0
PRINCIPAL PAYMENT	520	586	660	744	8 3 8	0	0
INTEREST PAYMENT	374	308	2 34	150	55	0	0
TOTAL: PRINCIPAL AND INTEREST	894	894	894	894	894	0	0
PRETAX PROFITS	-1110	-1630	-1630	1162	1162	2056	2056
TAXES	-1269	- 892	- 857	466	509	946	946
AFTERTAX PROFITS	159	- 739	- 773	697	653	1110	1110
CASH FLOW	- 917	2054	2020	697	653	1110	1110
CUMULATIVE PROFITS	159	- 580	1440	2137	2790	3900	5011
CUMULATIVE CASH FLOW	- 917	1136	3156	3853	4506	5616	6727

SIMPLE PAYBACK (YR) INTERNAL RATE OF RETURN (%) 3.26

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TABLE 26

RC-1 BOTTOMING CYCLE SYSTEM - CASH FLOW PROJECTION

COST OF MONEY	128
CORPORATE TAX RATE	468
INVESTMENT TAX CREDIT	78
ESCALATION RATE	08
DIESEL FUEL PRICE/GAL	\$1.20
ANNUAL MILEAGE/TRUCK	100,000
HARDWARE USEFUL LIFE (YR)	7
OPERATION AND MAINTENANCE COST/MIL	\$0.011
ORCS COST	\$8378.00
ORCS COST DIFFERENTIAL	\$6698.00
ORCS POWER OUTPUT (HP)	56
COST/HP ABOVE BASE ENGINE POWER	\$30.00
MPG BASE DIESEL ·	5.66
MPG ORCS/DIESEL	6,65
DEPRECIATION RATE/YR	3 8
DEBT	1008
LOAN TERM (MON)	60

YEAR	1	2	3	4	· 5	6	7
ORCS COST	0	0	0	0	0	0	0
ORCS COST DIFFERENTIAL	0	0	0	0	0	0	0
INVESTMENT TAX CREDIT	586	0	0	0	0	0	0
DEPRECIATION	2793	2793	2793	0	0	0	0
FUEL SAVINGS	3156	3156	3156	3156	3156	3156	3156
OPERATION AND MAINTENANCE COST	1100	1100	1100	1100	1100	1100	1100
OPERATING SAVINGS	2056	2056	2056	2056	2056	2056	2056
LOAN PRINCIPAL/END	5658	4486	3165	1677	0	0	0
MONTHS ON NOTE	48	36	24	12	0	0	0
PRINCIPAL PAYMENT	1040	1172	1 3 2 1	1488	1677	0	0
INTEREST PAYMENT	748	616	467	300	111	0	0
TOTAL: PRINCIPAL AND INTEREST	1788	1788	1788	1788	1788	0	0
PRETAX PROFITS	-1484	- 2524	- 2524	268	268	2056	2056
TAXES	-1613	-1444	-1376	-14	72	946	946.
AFTERTAX PROFITS	129	-1080	-1148	283	196	1110	1110
CASH FLOW	1882	1713	1644	283	196	1110	1110
CUMULATIVE PROFITS	129	- 951	694	976	1172	2283	3393
CUMULATIVE CASH FLOW	1882	3594	5239	5522	5718	6828	7938
SIMPLE PAYBACK (YR)	3. 26						
INTERNAL RATE OF RETURN (8)	N/A						

INTERNAL RATE OF RETURN (%)

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3. FUTURE PLANS

With the increased efforts to improve the efficiency of diesel engine prime movers, especially with the adiabatic engine concept, parallel efforts for the development of waste-heat recovery systems must be continued.

As the high-temperature problems (i.e., high-strength ceramic engine components) and other technical barriers are solved, the highefficiency adiabatic diesel engine will become an accepted reality. With that reality comes the characteristic of increased exhaust-gas temperatures of the adiabatic engine. All advanced bottoming cycle systems and associated technology problems should, therefore, be addressed with the same urgency as those barriers facing the adiabatic engine.

Investigations of high-temperature, advanced, waste-heat utilization systems should include those that have the potential for even greater fuel savings than those already considered. The high-performance RC-1 Rankine bottoming cycle is one such system.

3.1 RC-1 HIGH-PERFORMANCE CYCLE

Waste-heat recovery from the higher temperature exhaust gas of the adiabatic diesel engine and conversion to useful power can be maximized by implementing the RC-1 High-Performance Cycle (RC-1 HPC). The operation of the RC-1 HPC is a modified version of the simple Rankinecycle and is diagrammed in Figure 23. The modifications include: the use of two (2) turbines instead of one (1) or, alternatively, a two-stage, single turbine expander; a second vapor-generator in place of the regenerator located after the first turbine expander; and a working fluid flow split from the system's feedpump into two (2) paths (see state points 5 and 7 in Figure 23). The cycle modifications also include regeneration using the hot vapor from the second turbine expander to preheat the liquid working fluid entering the primary vapor-generator. These modifications are practical with the RC-1 working fluid because it has the ability to operate at a high system (turbine inlet) temperature of 850°F (potentially up to 1000°F), and because of the thermodynamic characteristic of the fluid to maintain a high vapor temperature after expansion through a turbine. Temperature drops of only 125° to 150°F are typical. Thus, the (first) turbine's vapor exhaust can be used as a heat source for a second Rankine-cycle. With the substitution of a vapor-generator heat exchanger for the regenerator of the simple cycle and the division of the feedpump fluid flow rate, a second closed loop (identified by state points 7, 8, 9, and 10 in Figure 23) can be accommodated.

The high-performance cycle (HPC) has not been evaluated in the depth indicated by the simple cycle (SC) conceptual design study results



Figure 23. Schematic of an Organic Rankine High Performance Cycle (HPC) System with Internal Heat Regeneration

A-6679

which are the subject of this report. However, preliminary comparison of performance estimates for the two (2) cycle configurations indicates (under idealized cycle conditions) the HPC has a potential to produce a compound engine BSFC that is 3 to $3\frac{1}{2}$ percent lower than that produced by the SC over a diesel engine exhaust temperature range from 1000° to 1600°F. This BSFC potential for the HPC translates into power improvements from 15 to 25 percent over the SC. While both systems are operating at 1000 psia, the organic operating temperature for the HPC is 900°F compared to the SC at 750°F.

3.2 TECHNOLOGY DEVELOPMENT AREAS

The majority of the technology barriers associated with Rankine-cycle systems used to bottom diesel engines with moderate exhaust gas temperatures between 650° to 1000°F (i.e., Fluorinol Truck ORCS) have been identified and dealt with for the most part. Except for the problem of heat exchanger fouling from the exhaust gas of diesel engines that burn No. 2 fuel oil, the majority of development work required on these systems, such as improvement in reliability, are solvable engineering problems.

These higher temperature, Rankine, bottoming systems, especially the RC-1 HPC at organic fluid operating temperatures approaching 1000°F, introduce new areas for technology development, as does any higher temperature operating system (e.g., adiabatic diesel engine, gas turbine). Some of the technology barriers that must be addressed for successful development of these high-temperature, high-performance Rankine bottoming cycle systems are identified below.

- Heat Exchanger Fouling This problem, as mentioned above, is one that applies to any bottoming cycle system that requires a heat exchanger to capture the exhaust gas waste heat of the diesel engine. As with the conventional diesel engine, this technology barrier is also common to the adiabatic diesel engine applications.
- Organic Working Fluid Thermal Stability This report has described the tests performed on the RC-1 and the results that show this fluid is capable of operating at temperatures up to 900°F. If systems such as the RC-1 HPC are to be considered as viable waste-heat recovery systems, then more organic fluid studies and testing must be accomplished for temperatures in excess of 1000°F. Systems and methods of fluid conditioning must also be investigated and developed if fluid degradation occurs at these elevated temperature levels.

High-Temperature Seal Development - Seals presently used on the turbine shaft (Fluorinol Truck ORCS) are of a double-faced, buffered chamber design with a lubricating oil as the buffer fluid. These seals prevent the entry of air into the working fluid, but do permit a small amount of the buffer fluid to pass into the working fluid space. The use of oil in the buffer chamber cools and lubricates the seal rubbing faces. There is no problem with the small amount of oil getting into the system because it is chemically compatible with the Fluorinol working fluid and thermally stable at the working fluid temperatures.

The same scheme cannot be used with RC-1 working fluid because the lubricating oil would decompose at RC-1 working fluid temperatures. Therefore, high-temperature seal cooling and lubrication, proper seal materials, and seal life with RC-1 are areas that require further technical investigation and development.

REFERENCES

- DiBella, F., DiNanno, L., and Koplow, M.: Laboratory and On-Highway Testing of Diesel Organic Rankine Compound Long-Haul Vehicle Engine. SAE Publication: 830122.
- (2) Standard Method of Test for Neutralization Number by Color-Indicator Titration. ASTM Designation: D974-64 (Reapproved 1968).
- (3) Miller, D.R. et al.: Optimum Working Fluids for Automotive Rankine Engines. Volumes I, II, III, and IV, Report Nos. APTD-1563 through APTD-1566, Monsanto Research Corporation. St. Louis, Missouri, June 1973.

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APPENDIX A

ANALYSIS REPORT OF RC-1 FLUID SAMPLE (800°F) BY CAMBRIDGE ANALYTICAL ASSOCIATES

1. INTRODUCTION

On September 14, 1983 sample IV-L-9/13/83-2L (CAA ID 83-8105) was submitted for GC/MS analysis of possible contaminants. A standard sample of RC-1 (CAA ID 83-8106) was also analyzed for comparison.

2. EXPERIMENTAL

The analytical conditions employed are summarized in Table 1.

3. RESULTS AND DISCUSSION

Figure 1 shows the reconstructed gas chromatogram (RGC) for both samples. As can be seen, they are identical. The large, saturated peak is due to pentafluorobenzene and hexaflourobenzene, the two major components.

The peak appearing at scan 60 is due to pentafluorochlorobenzene. The mass spectrum of this compound is given in Figure 2A. This peak was present in both samples at the same approximate concentrations. Four other minor components were also detected in both the sample and the standard. Figures 2B and 3 give the mass spectra of two of these peaks. A manual search of the NBS reference library spectra of 32,000 compounds gave no matches. The mass spectra of the peaks indicate that they are fluorinated compounds (characterized by an $(M^+ - 19)$ ion), and both aromatic and aliphatic in nature, as the molecular ion is of medium intensity. Alkanes would give a much weaker molecular ion, and aromatics would give a much more intense one. The four minor components had molecular weights of 212, 218, 236 and 262.

In conclusion, no impurities were found in the sample that were not detected in the standard.

Table 1

ANALYTICAL CONDITIONS

SAMPLE TYPE:	Liquid
INTRODUCTION:	on column injection
INSTRUMENT:	Finnigan 9500GC/3200MS/6000DS
GC COLUMN:	6ft x 1/4" OD OV-101 packed column
INJECTOR TEMPERATURE:	220 [°] C
INITIAL TEMPERATURE:	35 ⁰ C
INITIAL TIME:	2 minutes
TEMPERATURE PROGRAM:	35 ⁰ C to 300 ⁰ C at 12 ⁰ C/minute
FINAL TEMPERATURE:	300 ⁰ C
FINAL TIME:	12 minutes
MASS RANGE SCANNED:	41 to 350 amu
SCAN RATE:	3 seconds per scan

No.7







AMP .: 00036864

unidentified compound

APPENDIX B

ANALYSIS REPORT OF RC-1 FLUID SAMPLE (900°F) BY CAMBRIDGE ANALYTICAL ASSOCIATES

INTRODUCTION

On September 29 and November 9, 1983 the following samples were submitted for GC/MS analysis of possible contaminants.

Thermoelectron ID	CAA ID	Sample Type
IV-G-9/29/83-1G	8308428	gas
III-L-11/16/83-1L	8309123	liquid
IV-L-11/7/83-1L	8309124	gas

EXPERIMENTAL

The analytical conditions employed are summarized in Table 1.

RESULTS AND DISCUSSION

Comparisons were made between this analysis and previous analyses done at CAA (see CAA Report 83-788). Sample IV-G-9/29/83-1G was of insufficient pressure to get an intense chromatogram. Only pentafluorobenzene and hexafluorobenzene were detected in this sample.

Sample IV-L-11/7/83-1L had a large peak that eluted before the fluorinated benzenes, (see Figure 2). The compound had no matches against the NBS library of reference compounds. It was tentatively identified as molecular weight 186 with a possible structure of perfluoro-n-propanol (see Figure 3). Sample II-L-11/16/83-1L had pentafluorochlorobenzene, as seen in

previous analyses of the liquid (see Figure 4). This compound was not present in the gas samples. Two other major peaks were detected (see Figure 5 and 6). They appear to be fluoroalkanes of molecular weights 348 and 338. Fluoroalkanes of molecular weight 316 and 334 were also detected. Sample IV-L-9/13/83-2L was reanalyzed to make sure that these compounds weren't missed in previous analyses due to altered chromatographic condition. They were not detected (see Figure 7).

All compounds were present at low levels (estimated to be parts per million) except for the perfluoro-n-propanol in sample III-L-11/16/83-1L. This compound is estimated to be present at low percent levels. Hydrofluoroic acid would not be detected in this analysis.

Table 1

ANALYTICAL CONDITIONS

SAMPLE TYPE: liquid/gas

INTRODUCTION: liquid - direct injection/gas - loop injection

INSTRUMENT: Finnigan 3200 GC/MS

GC COLUMN: 6 ft x 1/4" OD OV-101 packed column

INJECTOR TEMPERATURE: 220°C

INITIAL TEMPERATURE: 35°C

INITIAL TIME: 2 minutes

TEMPERATURE PROGRAM: 35°C to 300°C at 12°C/minute

FINAL TEMPERATURE: 300⁰C

FINAL TIME: 12 minutes

MASS RANGE SCANNED: 41 to 350 amu

SCAN RATE: 3 seconds per scan



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APPENDIX C

RC-1 THERMODYNAMIC PROPERTIES

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SATURATION

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Р	Т	VL	VV	ΗĹ	HV
(PSIA)	(°F)	(FT^3/LB)	(FT ³ /LB)	(BTU/LB)	(BTU/LB)
.23	.00	.9217E-02	.12288 03	7054E 01	.88968 62
.19	5.00	.92695-02	.9163E 02	6067E 01	.8957E 02
-27	10.00	.9336E-02	.8454E 02	4940E 01	.9027E 02
.34	15.00	.94938-92	.7746E 02	3814E 01	.9098E 02
•42 57	20.00	.9470E-02 0502E-00	-7937E 96 SAGOR GO	25571 01 	.7167E 00 Goode (62
- 35 · 40	20.00		- 3402E 02 - 5002E 02	- SORE DI	1204E OE 12047E D2
.61	30.00 35.00	.96678-92	.4603F 02	. 6578F AA	-90 1000 -9279F 02
.73	40.00	.9737E-02	.42035 02	.1819E 01	.9452E 02
.67	45.00	.97928-02	.3308E 02	.2875E 01	.9520E 02
.85	50.00	.9868E-02	.3074E 02	.4071E 01	.9594E 02
1.04	55.00	.9943E-02	.2840E 02	.5267E 01	.96698 02
1.23	60.00	.10026-01	.2606E 02	.6464E U1	.9743E U2
1.14	65.00	.1008E-01	.2090E 02 1989E 03	.7360E 01 07096 01	.9807E 08 900985 89
1.40 1.20	78.00 25 00	10205-01	1911F 02	.01295 91 10025 02	.700JE UE 994APE 82
1.98	80.00	.10328-01	.1669E 92	.1125E 02	. 1004E 03
1.89	85.00	.1038E-01	.1368E 02	.1238E 02	.1011E 03
2.29	90.00	.1047E-01	.1279E 02	.1364E 02	.1019E 03
2.69	95.00	.1055E-01	.1190E 02	.1490E 02	.1827E 03
3.10	100.00	.1064E-01	.1101E 02	.1617E 02	.1035E 03
3.60	195.88	.1071E-01	.9170E 01	.1734E 02	.1043E 03
3.00 4 15	110.00	.1080E-01	.8533E Ul	.1863t 82 (aaar aa	.1051E 03 1050E 00
4.10	120.00	10926-01	2457F 01	-1773E 86 2123F 82	1005E 03
4.66	125.00	.11056-01	.63956 01	.2241F 02	. 1075E 03
5.44	130.00	.1115E-01	.5928E 01	.2375E 02	.1083E 03
6.23	135.00	.1125E-01	.5551E 01	.2508E 02	.1091E 03
7.01	140.00	.1135E-01	.5173E 01	.2642E 02	.1099E 03
6.97	145.00	.1142E-01	.4429E 01	.2766E 02	.1106E 03
8.03	150.00	.1153E-01	.4174E 01	.29026 02	.1115E 03
9.09	100.00	11046-01	- 3919E 01 9664E 60	.3937E 96 01766 80	.1183E 03 1191E 09
10.13	165.00	.11826-01	.3004C 01	.3303F 02	.1131E 03
11.59	170.00	.1194E-01	.2997E 01	.3443E 02	.1147E 03
12,99	175.00	.1206E-01	.2821E 01	.3583E 02	.1155E Ø3
14.39	180.00	.1217E-01	.2645E 01	.3723E 02	.1164E Ø3
14.52	185.00	.1225E-01	.2313E 01	.3852E 02	.1173E 03
16.35	190.00	.12388-01	.2189E 01	.3996E 02	.1181E Ø3
18.17	195.09	.1201E-01	.2065E 01	.4137E 02 4000E 00	.1189E 03 1100E 00
20.00 20 30	200.00	.12040-01 .1273F-01	17125 01	.42000 00 .4416F 02	.1176E 00 .1206F 03
22.63	210.00	.1287E-01	.1624E 01	.4563E 82	.1215E Ø3
24,97	215.00	.1301E-01	.1535E 01	.4711E 02	.1223E 03
27.31	220.00	.1315E-01	.1446E 01	.4858E 02	.12320 03
27.79	225.00	.1324E-01	1285E 01	.4994E 02	.1241E 03
30.75	230.00	.1339E-01	.1221E 01	.5145E 02	.1249E 03
33.71	235.00	.1355E-01	.1156E 01	.5296E 02	.1258E 03
30.00	240.00	1071E-01 1091E-01	.1072E 01 4222E 00	.04471 82 55026 80	1205E 93 1024E 80
41.12	250.00	.1398F-01	.92955 00	.5566 86 .5741F 82	. 1222F 03
44.80	255.00	.1415E01	.8818E 00	.5896E 82	.1292F M3
48.48	260.00	.1433E-01	.8340E 00	.6051E 02	.1300E 03
49.61	265.00	.1443E-01	.7506E 00	.6193E 02	.1308E 03
54.14	278.00	.1463E-01	-7149E 00	.6352E 02	.1316E G3
58.67 20 00	275.00	.1483E-01	.6789E 00	.6512E 02	.1324E 03
64.60 44 70	200.00	15145-01	-6431E UU Setar Du	.5671E 82 20176 00	.13338 83
204.10 20.29	290.00	.15378-01	.5542F 00	100171 02 20171 02	-13421 03 19501 00
75.81	295.00	1559E-01	.5270E NA	.07010 ос .7145F ир	- 1000E 900 12590E 900

SATURATION

Р	т	VL	VV	HL	HV
(PSIA)	(°F)	(FT ³ /LB)	(FT³/LB)	(BTU/LB)	(BTU/LB)
81.32	300.00	.15826-01	.4998E 00	.73698 08	2 .13678 93
83.45	305.00	.15938-01	.4539E 00	.7456E 0a	e .1377E 03
99.09	310.00	.16208-01	.4329E 00	.7625E 0a	e .1384E 03
96.74	315.00	.16468-01	.4119E 00	.77958 08	e .1392E 03
193.38	320.00	.1673E-01	.39095 001	.7965E 00	2 .1400E 03
106.19	325.00	.1684E-01	.35628 00	.8114E 0a	e .1408E 03
114.11	330.00	.17166-01	.3398E 00	.8290E 00	2 .1416E 03
122.04	335.00	.1748E-01	.3234E 00	.8467E 08	2 .1423E 03
129.96	340.00	.1780E-01	.30708 00	.86448 00	2 .1431E 03
133.53	345.00	.17896-01	.2805E 00	.87975 00	2 .1441E 03
142.91	350.00	.18296-01	.2675E 00	.89815 0	2 .1448E 03
152.28	355.00	.1868E-01	.2545E 00	.9166E 00	2 .1455E 03
161.66	360.00	.19088-01	.2415E 00	.9351E 6:	· 1462E 03
166.14	365.00	.1912E-01	.2210E 00	.9490E 0	2 .14716 03
177.15	370.00	.1964E-01	.2105E 00	.9690E 00	2 .14788 03
188.16	375.00	.2015E-01	.20005 00	.98906 00	2 .1484E 63
199.17	389.00	.2067E-01	.1895E 00	.1009E 0	3 .1490E 03
204.59	385.00	.2054E-01	.1738E 00	.1622E 0	3 .15998 93
217.46	390.00	.2127E-01	.1650E 00	.1044E 0	3 .15050 03
230.34	395.00	.2199E-01	.1562E 00	.1067E 0	3 .1510E 03
243.21.	400.00	.2272E-01	.1475E 00	1089E 0	3 .15150 03
248.05	405.00	.2178E-01	.1373E 00	.10968 00	3 .1533E 03
263.60	410.00	.2307E-01	.12968 00	.11238 0.	3 .1533E 03
279.16	415.00	.2435E-01	.1207E 00	.1151E B	3 .1534E 03
294.71	420.00	.2563E-01	.1124E 00	.1178E 0	3 .15355 03
396.22	425.00	.2226E-01	.1159E 00	.1164E Q	3 .1649E 03
323.12	430.00	.25100-01	.1007E 00	.1896E 0	3 .1612E UG
340.02	435.00	.2793E-01	.91396-01	.1247E 0	3 .15750 00
356.92	440.00	.3077E-01	.79116-01	.1288E 00	3 .1538E 03
402.73	445.00	.3984E-01	.3984E-01	.1420E 00	3 .1420E 03

STOP 0

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SUPERHEATED VAPOR						
Р	т	н	S	V	СР	
(PSIA)	(°F)	(BTU/LB)	(BTU/LB-°F)	(FT ³ /LB)	(BTU/LB-°F)	
5,69	± 0.00	.64648 01	.1461E-01	.10028-01	.957	
5.64	80.NG	.11258 00	.24658-01	.10328-01	.984	
5.00	100.06	.1617E 02	.34408-01	.1664E-01	1.012	
5.00	120.00	.2123E 02	.4389E-01	.10988-01	1.038	
5.00	140.00	1898E 03	.1963E 00	.7239E 01	.172	
5.59	166.09	.1134E 03	.2018E 00	.7491E 01	. 175	
5.69	188.00	.1169E 03	.2073E 00	.7742E 01	.178	
ે. તાન	200.00	.1205E 03	.21288 00	.7994E 01	.180	
5.60	220.00	.1242E 03	.2183E 00	.8241E 01	.185	
5.60	240.00	.1279E 03	.2238E 00	.8489E 01	.198	
5.00	268.08	.1318E 03	.2291E 00	.8737E 01	. 195	
5.60	230.00	.1356E 03	.2345E 00	.8985E 01	.200	
5.69	300.00	.1396E Ø3	.2398E 00	.9236E 01	.203	
5.00	320,00	.1437E 03	.2451E 00	.9488E 01	.205	
5.90	340.00	.1479E 03	.25938 00	.9734E 01	.208	
5.90	360.00	.1521E Ø3	.25568 00	.9980E 01	.210	
5.00	380.00	.1564E 03	.2607E 00	.10238 02	.215	
5.00	400.00	.1607E 03	.2659E 00	.1047E 02	.220	
5.00	428.08	.1651E 03	.2710E 00	.1072E 02	.225	
5.00	448.88	.1696E 03	.276GE 00	.1097E 08	.230	
5.68	460.00	.1742E 03	.2810E 00	.1122E 08	.232	
5.80	488.00	.17888 03	.2860E 00	.1146E 02	.235	
5.80	500.00	.1836E 03	.2910E 00	.1171E 08	.237	
5.69	220.00	.1883E 03	.2959E UO	.1196E 02	.249	
5.60	540.00	.1931E M3	.3097E 00	.12205 08	.242	
5.69	568,08	.19868 63	.30566 00	.12456 02	.245	
5.60	588.00	.2929E 03	.31945 00	.1270E 02	.250	
5.00	688,88	.2879E 83	.31512 00	.1894E 02	.255	
5.UU*	628.88	.2130E 03	.31985 00	.13196 08	.200	
5.00	649.00	.2181E 03	.3246E 00	.1343E 08	.255	
5.99	660.00	.2233E 03	.3272E 00	.1368£ 02	.257	
5.00	000.00	.8280E 03	.33388 88	.1393E 00	.259	
0.99	700.00	.2338E 03 	.3334E 80 34396 00	.14175 0s		
0.00 8 00	740.00	.COVIE DO Daard oo	.04000 00 04755 00		:	
્યત્ર દીધી મંદ્ર તેવાલ	140.00	.04400.00 	-0410E 00 0400E 00	.1400E 00 1401E 00	: • (20))) (27)3	
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	100.00 Vəbi Əbi		.00046 00 126696 88	1530F 05	L.C) 2223	
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្រ.១១ មុខាណ	545.55 266 66	2777F AR	20000000 202945 00	1613F 83	. ,000) jogq	
5.00 5.00	200 . 00 228 HB	2224F M2		16745 63		
5.111	900 . 00 900 00	229115 02	2824F 00	10 JC001. 20 JC001	· · · · · · · · · · · · · · · · · · ·	
5 111	928. GM	29498 83	.3866F 88	.1687F 08	· .200	
5.66	946.66	.3898F 93	SECTION AND	.1712F 02		
5.63	964 44	.3066F 03	.3950F 00	.1737F 02		
5.00	988.64	.3125E 03	.3991E 00	.1761E MA	297	
5.60	1000.00	.3184E 03	.4032E 00	.1786E 08	.390	
5.00	1929.00	.3244E 03	.4072E 00	.1810E 08	300	
5.00	1040.00	.3304E 03	.4113E 00	.1835E 08	.300	

P	T		S (DTU/LD SE)	V (FT3/(D)	СР
(PSIA)	(**)	(BIU/LB)	(BIU/LB-*F)	(FIS/LB)	(BIU/LB-°F)
111, 411	F161 6161	. A4A4F (4)	.:4618-01	.19928-01	
19.00	860 660	.1125E 62	2465E-01	10028-01	.984
19.40	1 6161 6161	.1617F 92	.3440E-01	1064E-01	1.012
14.80	1291.66	. 123E 02	.4389E-01	.1098E-01	1.033
10.00	148.86	2642E 82	.5316E-01	.1135E-01	1.868
10.00	160.00	.1131E 03	.1937E 00	.3698E 01	180
16.60	186.66	TIMEF MR	.1992E 00	.3825E 01	.182
10.00	200.00	1202E 93	.2048E 00	.0953E 01	.185
11 110	2291.00	.1239E 93	.2102E 00	.4080E 01	187
1.1.1.1.1	240.00	.1277E 03	.2157E 00	.4807E 01	.199
10.00	266.66	.1316E 93	.2211E 08	.43338 01	. 195
10.00	230.00	.1355E 93	2265E 00	.4458E 01	.200
10.00	399.99	1395E 03	.2318E 00	.45848 01	205
10.00	32 9 .99	.1435E 03	.2371E UU	.4710E 01	216
18,00	340.00	.1477E U3	.2423E 00	.4835E 01	. 212
10.00	360.00	.1519E 60	.2476E 00	.4960E 01	. 215
16.00	330.00	1563E 03	.2527E 00	.50068 01	.217
10.00	400.00	.1606E 03	.2579E 00	.52116 01	. 220
10.00	420.00	.1651E 03	.2630E 00	.5335E 01	.225
16.00	440.00	.1695E U3	.2681E 00	.5459E 01	.236
10.00	468.99	1741E 03	.2731E 00	.55838 01	.233
16.00	486.00	.1787E 03	.2781E 06	.57088 01	.235
10.00	500.00	.1834E 03	.2830E 00	.58328 01	.237
10.00	520.00	.1882E 03	.28795 00	.5956E 01	. 249
18.00	549.00	.1930E 03	.2928E 00	.6080E 01	.243
10.00	560.00	.1979E 03	.29778 00	.62635 01	.245
10.00	588.88	.2828E 83	.3025E 00	.8328E 0:	.258
10.53	666.66	.2078E 03	.3072E 00	.6452E 81	.255
19.99	628.99	.2129E Q3	3119E 00	.6577E U	.257
10.00	640.00	.2180E U3	.3166E 00	.6792E 91	.260
18.69	EEU.UU	.2232E V3	.32138 00	.60268 0	1.268
18.QS	680.00	.2284E 03	.32598 00	.6949E 0	.260
10.00	ିମ୍ମାରି ପୂର୍ବ	.2337E 93	.3305E 00	.7072E 0	.265
10.00	729.00	.23905 03	.33515 00	.7194E 0:	1.279
18.00	740.00	.2444E 03	.3396E 00	.7317E 0:	1 .270
10.00	760.00	.2498E 93	.3441E 00	.7440E 0	1 .279
10.00	788.09	.2553E 63	.3485E 00	.7563E 6:	.272
16.00	888.00	.26088 00	.3529E 00	.7686E U	1 .275
10.00	820.00	.2663E 93	.3573E 00	.7811E 9	1 .289
19.99	មុនមិន ស្រុង	.27195 03	.3617E 00	.7937E 8	.285
16.60	860.00	.2776E 83	.3660E 00	.8060E 0:	.285
16.66	880.00	.2833E 93	.3703E 60	.8183E 8:	.235
10.00	900.00	.2891E 63	.3745E 00	8394E 01	.287
10.00	920.00	.2348E 03	.STERE UU	.8425E 0:	.296
10.00	940.00	.SUUGE US	.38298 00	.8549E 01	. 292
10.00	950.00	.3865E 83	.38715 80	.8673E 81	.295
10.00	788.88	.3124E 83 	.3712E 80	.8797E 01	.297
18,89 18,199	1000.00	.3183E 83 33400 A	.3953E 88	.8921E 01	.399
1년•년년 1년•년년	XUEU.UM NGRA GG	.0040E 83 	.3773E 88 Aboat oo	.9043E U	
비행 비행인	* 아파안 * 양민	UC UC	.40346 80	. FIBBE 01	L .UUU

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́ Р	Т	н	S	· -	СР
(PSIA)	(°F)	(BTU/LB)	(BTU/LB-°F)	(FT ⁻³ /LB)	(BTU/LB-°F)
····,		(,		• • • • • • •	
15.66	6.6	6.46.4F B1	14616-01	19925-01	್ರವ್ಯತ
19 141	288 HB	11258 02	.14010 01 24650-01	1632F-01	 ଜାନାଣ
15.00	100.00	16126 00	2440F-01	10445-01	
15 66	128.00	0100E 80	17990F-01	10090-01	1 1000
15 86	100.00	-01000 00 02305 00	- 40000 01 - 40120-01	11056.01	. 1.000 1 Reú
1000 UN 1900 UN	160.00	0172L 00	.00102 01 .00102 01	11745-01	. 1.000 1 Gua
and a state	100.00	-01(DE BE -0700E 80	.02000-01 71100-01	10170-01	
	100.000 1000 - 000	10120E-UC 10008E-000		- ICI/E-01 	. 1.1CU 10F
1.1.1 + 5353 1.1.2 - 1.2.12		.18008 03	-2000L 00	COCIE - OI 	. е 4.0.504 Албайа
10.55		.16376 05 10786 00	.2830 <u>2</u> 88 	1911 1911 1911 1919 1911 191	· • 上之初 《山田
10.00	240.00	.1270b.93	-CIIOE 00	.COUIL UI	
10.00	200.00 200 00	.1314 <u>L</u> 93	21046 00 Senter da	LOGIE UI	171
10.00	280.00	.1333L 83	.2218L 80	-CTICL UI	200
11년 - 12년 11년 - 12년	000.00	.13738 80	.CC/12 00	.3807E 01	. 200
10.99	320.00	.14348 03	.23246 00	.3143E 01	200
10.99	349.00	.14/6t 03	.23771 90 	.32398 01	
10.00	300.00	.10188 03	.24276 80	.3310t UI	
13.00	330.00	.1361E 03	.24811 60	.34905 01	220
15.00	400.00	.16048 03	.2033t 80	.3484E U.	
15.69	420.00	.16495-03	.2584E UU	.3569E 01	
15.00	446.66	.1694E US	.2634E 60	.3653E UI	. 225
15.69	460.00	.1749E 03	.2684E UU	.37375 01	.230
12.00	489.00	.1786E Ø3	.2734E 88	.3855E N	.235
15.00	590.09	.1833E 03	.2784E 00	.3906E M1	.240
15.99	528.98	.1880E 00	.28335 00	.3990E 01	.245
15.60	549.00	.1929E 03	.2882E 00	.4074E 0	.245
15,60	560.00	.1978E U3	.2930E 00	.4158E B	.245
15.00	580.00	.2027E 03	.2978E 00	.4242E 0	.250
15.00	668.66	.2077E 03	.3826E 00	.4325E 0	.255
15.00	620,00	.2128E 03	.3973E 60	.4469E 0	.250
15.00	649.00	1.2179E 03	.3120E 00	.4493E 0	1.260
15.99	668.00	.2231E 03	.3167E 00	.4576E 0	1.263
15.00	680.00	.2283E 03	.3213E 00	.4660E10	.265
15.00	700.00	.2336E 03	.3259E 00	.4743E 0	.268
15.00	720.00	.2389E 03	.3305E 00	.4826E 0	1 .270
15.00	740.00	.2443E 03	.3350E 00	.4909E 0	1.273
15.00	760.00	.2497E 03	.3395E 00	.4992E 0	1 .275
15.00	780.00	.2552E 03	.3439E 00	.5076E 0	1 .278
15.60	899.00	.2607E 03	.3483E 00	.5159E 0	1.280
15.00	820.00	.2663E 03	.3527E 00	.5242E 0	1 .289
15.00	840.00	.2719E 03	.3571E 00	.5326E 0	1.289
15.00	860.00	.2775E 03	.3614E 00	.5409E 0	1.285
15.00	886.00	.2832E 03	.3657E 00	.5492E 0	1.290
15.00	999.99	.2890E 03	.3700E 00	.5575E 0	1 .298
15.00	920.00	.2948E 03	.3742E 00	.5658E 0	1.598
15.00	940.00	.3006E 03	.3784E 00	.5742E 0	1 .293
15.00	960.00	.3064E 03	.3825E 00	.5825E 0:	.295
15.00	980.00	.3123E 03	.3866E 00	.5908E 0	.295
15.00	1880.08	.3183E Ø3	.3907E 00	.5991E 0:	.295
15.00	1020.00	.3243E Ø3	.3948E 00	.6073E 0	.298
15.00	1646.00	.3303E 03	.3988E 00	.6156E 0:	.300

	P (PSIA)	T (%E)	H (BTU/LB)	S (BTU/LB-°F)	V (FT ³ /I B)	CP
	(FJIA)	(1)	(010720)		(,,	
$\begin{array}{cccccccccccccccccccccccccccccccccccc$		641. LIU	. Бабағ ит	.1461E-01	.10028-01	. 957
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	84.00	1125E 02	2465E-01	.1032E-01	.984
$\begin{array}{llllllllllllllllllllllllllllllllllll$	ວິນີ. ນທີ	100.00	.1617E 02	.3440E-01	.1064E-01	1.012
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	120.00	.2123E 02	.4389E-01	.1098E-01	1.038
$ \begin{array}{llllllllllllllllllllllllllllllllllll$	29.ពម	145.00	.2642E 02	.5316E-01	.1135E-01	1.068
$\begin{array}{llllllllllllllllllllllllllllllllllll$	29.99	160.00	.3176E 02	.6223E-01	.1174E-01	1.094
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	180.00	.37238 02	.7112E-01	.1217E-01	1.120
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	200.00	.1196E 03	.1970E 00	.2078E 01	. 182
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	ខូមិ.ម៉ូម៉	229.99	.1235E 03	.2025E 00	.2148E 01	. 187
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	240.00	.1273E Ø3	.2079E 00	.2218E 01	192
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	260.00	.1312E 03	.2133E 00	.2287E 01	. 196
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	29.00	229.00	.1351E 03	.2187E 00	.2356E 01	. 200
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	366.66	.1391E 03	.2249E 00	.2426E 01	. 204
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	029.90	.1432E 03	.2294E 66	.2495E 01	. 298
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	340.00	.1474E 03	.2346E 00	.25638 61	.211
20, 00 $330, 00$ $.1559E$ 03 $.2451E$ 00 $.2768E$ 01 $.226$ $20, 00$ $420, 00$ $.1642E$ 03 $.2553E$ 04 $.236E$ 01 $.225$ $20, 00$ $420, 00$ $.1642E$ 03 $.253E$ 04 $.2971E$ 01 $.223$ $20, 00$ $420, 00$ $.1692E$ 03 $.2664E$ 09 $.2971E$ 01 $.231$ $20, 00$ $420, 00$ $.1732E$ 03 $.2654E$ 09 $.309E$ 01 $.205$ $23, 00$ $500, 00$ $.1832E$ 03 $.2754E$ 09 $.3106E$ 01 $.242$ $20, 00$ $520, 00$ $.1832E$ 03 $.2803E$ 09 $.3241E$ 01 $.242$ $20, 00$ $540, 00$ $.1976E$ 03 $.2994E$ 00 $.3275E$ 01 $.248$ $20, 00$ $540, 00$ $.1976E$ 03 $.2948E$ 00 $.3775E$ 01 $.248$ $20, 00$ $560, 00$ $.2076E$ 03 $.2948E$ 00 $.3775E$ 01 $.248$ $20, 00$ $660, 00$ $.2172E$ 03 $.3043E$ 00 $.3775E$ 01 $.265$ $20, 00$ $660, 00$ $.2172E$ 03 $.3229E$ 00 $.3716E$ 01 $.265$ $20, 00$ $660, 00$ $.2238E$ 03 $.3275E$ 00 $.3716E$ 01 $.268$ $20, 00$ $740, 00$ $.2338E$ 03 $.3275E$ 00	CU.UU	369.00	.1516E 03	53495 66	.2632E 01	.215
20,00 $40,00$ $1602E$ 03 $253E$ 00 $2236E$ 01 225 $20,00$ $426,00$ $1647E$ 03 $253E$ 00 $2236E$ 01 2231 $20,00$ $460,00$ $1739E$ 03 $2654E$ 00 $2971E$ 01 2231 $20,00$ $460,00$ $1739E$ 03 $2254E$ 00 $2971E$ 01 2231 $20,00$ $506,00$ $1832E$ 03 $2754E$ 00 $3039E$ 01 2239 $20,00$ $526,00$ $1832E$ 03 $2852E$ 00 $3174E$ 01 242 $20,00$ $540,00$ $1976E$ 03 $2932E$ 00 $3274E$ 01 242 $20,00$ $560,00$ $1976E$ 03 $2934E$ 00 $3372E$ 01 248 $20,00$ $560,00$ $1976E$ 03 $2948E$ 00 $3372E$ 01 248 $20,00$ $560,00$ $297E$ 03 $3043E$ 00 $359E$ 01 258 $20,00$ $560,00$ $297E$ 03 $3043E$ 00 $3576E$ 01 260 $20,00$ $640,00$ $2178E$ 03 $3091E$ 00 $3710E$ 01 265 $20,00$ $640,00$ $2178E$ 03 $3237E$ 04 $3777E$ 01 265 $20,00$ $640,00$ $2178E$ 03 $3275E$ 00 $3710E$ 01 273 $20,00$ $760,03$	50.00	380.00	.1559E 03	.2451E UU	.2790E MI	. 2211
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	400.00	.1602E 03	.2563E 60	.2768E 01	.225
$\begin{array}{llllllllllllllllllllllllllllllllllll$	59.99	426.00	.1647E US	.25538 88	.28365 01	.226
20, 00 $480, 00$ $1734E$ 03 $2764E$ 00 $3039E$ 01 225 $20, 00$ $506, 00$ $1832E$ 03 $2754E$ 00 $3106E$ 01 225 $20, 00$ $520, 00$ $1832E$ 03 $2754E$ 00 $3174E$ 01 242 $20, 00$ $520, 00$ $1928E$ 03 $2852E$ 00 $3241E$ 01 242 $20, 00$ $540, 00$ $1928E$ 03 $2932E$ 00 $3375E$ 01 243 $20, 00$ $580, 00$ $1976E$ 03 $2936E$ 00 $3375E$ 01 225 $20, 00$ $580, 00$ $2976E$ 03 $2936E$ 00 $3375E$ 01 258 $20, 00$ $660, 00$ $2127E$ 03 $3043E$ 00 $3509E$ 01 258 $20, 00$ $640, 00$ $2178E$ 03 $3137E$ 00 $3543E$ 01 262 $20, 00$ $660, 00$ $2230E$ 03 $3137E$ 00 $3643E$ 01 262 $20, 00$ $660, 00$ $2238E$ 03 $3232E$ 00 $3714E$ 01 268 $20, 00$ $740, 00$ $2238E$ 03 $3229E$ 00 $3777E$ 01 268 $20, 00$ $740, 00$ $2442E$ 03 $3320E$ 00 $3910E$ 01 273 $20, 00$ $780, 00$ $2238E$ 03 $3326E$ 00 $4110E$ 01 280 <td< td=""><td>20.00</td><td>449.99</td><td>.1682E 08</td><td>.26041 00</td><td>- 27304<u>5</u> 01 20072417 023</td><td></td></td<>	20.00	449.99	.1682E 08	.26041 00	- 27304 <u>5</u> 01 20072417 023	
20,00 $480,00$ $.1784E$ 63 $.2784E$ 60 $.3037E$ 61 $.239$ $20,00$ $520,00$ $.1879E$ 63 $.2832E$ 60 $.3174E$ 61 $.242$ $20,00$ $520,00$ $.1978E$ 63 $.2852E$ 60 $.3241E$ 61 $.242$ $23,00$ $560,00$ $.1978E$ 63 $.2931E$ 60 $.3375E$ 61 $.245$ $23,00$ $560,00$ $.1978E$ 63 $.2934E$ 60 $.3375E$ 61 $.245$ $20,00$ $580,60$ $.2626E$ 63 $.2948E$ 60 $.3375E$ 61 $.255$ $20,00$ $660,60$ $.2678E$ 63 $.3943E$ 60 $.3576E$ 61 $.260$ $20,00$ $649,60$ $.2178E$ 63 $.3641E$ 61 $.265$ $20,00$ $649,60$ $.2178E$ 63 $.3691E$ 60 $.3776E$ 61 $.260$ $20,00$ $649,60$ $.2178E$ 63 $.3643E$ 61 $.262$ $20,00$ $640,60$ $.2238E$ 63 $.3275E$ 60 $.3716E$ 61 $.263$ $20,00$ $740,60$ $.2335E$ 63 $.3275E$ 60 $.3977E$ 61 $.275$ $20,00$ $740,60$ $.2442E$ 63 $.3326E$ 60 $.3977E$ 61 $.275$ $20,00$ $740,60$ $.2442E$ 63 $.3326E$ 60 $.4116E$ 61 $.270$ $20,00$ $740,60$ <t< td=""><td></td><td>465.99</td><td>.1/38E-03</td><td>.25341 99 Storage ag</td><td>.2371E 01</td><td>L .EGl</td></t<>		465.99	.1/38E-03	.25341 99 Storage ag	.2371E 01	L .EGl
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.99	480.00	.1784b 63 tocor co	.2704E 00 0754E 00	.3937E 01 State at	L , <u>C</u> UUU 1
23.00 540.00 $.10770$ 03 $.28520$ 00 $.32410$ 11 $.245$ 23.00 560.00 19760 03 $.29910$ 00 $.33750$ 01 $.245$ 23.00 560.00 $.19760$ 03 $.29960$ 00 $.33750$ 01 $.251$ 20.00 660.00 $.20760$ 03 $.29960$ 00 $.34420$ 01 $.255$ 20.00 660.00 $.21270$ 03 $.30910$ 00 $.35760$ 01 $.256$ 20.00 660.00 $.21780$ 03 $.31370$ 00 $.36430$ 01 $.266$ 20.00 660.00 $.22300$ 03 $.31370$ 00 $.36430$ 01 $.265$ 20.00 660.00 $.22350$ 03 $.31270$ 00 $.37770$ 01 $.268$ 20.00 700.00 $.23350$ 03 $.32290$ 03 $.37770$ 01 $.268$ 20.00 740.00 $.23350$ 03 $.32290$ 03 $.37770$ 01 $.268$ 20.00 740.00 $.24420$ 03 $.323200$ 03 $.39770$ 01 $.275$ 20.00 740.00 $.24420$ 03 $.343200$ $.4943200$ 1.275 20.00 780.00 $.25510$ $.332500$ 00 $.397700$ 1.275 20.00 780.00 $.26660$ 03 $.344200$ $.41100$ 1.2800 20.00 820.00 $.2666$	<u> - 단</u> 단 - 단단 - 2013 - 1414	000.00 6 % 00	.:Cart 93 ::::::::::::::::::::::::::::::::::::	.21041 88 .200705 88		L .៥៤% E
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	- 2062 - 6262 - Sela - Gala	829.00 Ess 66	1000E 60 1000E 60	20000E 00 0050E 00		L 201400 L 20140
20,00 $580,00$ $2826E$ 63 $2948E$ 60 $3375E$ 61 251 $23,00$ $660,00$ $2876E$ 63 $2996E$ 60 $3442E$ 61 255 $28,00$ $620,00$ $2127E$ 63 $3043E$ 60 $3509E$ 61 258 $20,00$ $640,00$ $2127E$ 63 $3043E$ 60 $3576E$ 61 269 $23,00$ $660,00$ $2230E$ 63 $3137E$ 60 $3643E$ 61 262 $20,00$ $680,00$ $2230E$ 63 $3137E$ 60 $3643E$ 61 262 $20,00$ $680,00$ $2230E$ 63 $3137E$ 60 $3643E$ 61 262 $20,00$ $680,00$ $2238E$ 63 $3229E$ 60 $3710E$ 61 276 $20,00$ $740,00$ $2238E$ 63 $3229E$ 60 $3777E$ 61 276 $20,00$ $740,00$ $2442E$ 63 $3326E$ 60 $3977E$ 61 2773 $20,00$ $780,00$ $2496E$ 63 $3345E$ 60 $4474E$ 61 278 $20,00$ $780,00$ $2496E$ 63 $3442E$ 60 $4110E$ 61 278 $20,00$ $780,00$ $2662E$ 63 $3454E$ 60 $4177E$ 61 283 $20,00$ $840,00$ $2775E$ 63 $3535E$ 60 $4243E$ 61 293 $20,00$ $840,00$ </td <td>- 이번호 번호 - 이번호 테이</td> <td>540.00 540.00</td> <td>1920E NO 1972E NO</td> <td>2000CC 00 29001F 00</td> <td>RADELL DI</td> <td>L •도막교 1 231요</td>	- 이번호 번호 - 이번호 테이	540.00 540.00	1920E NO 1972E NO	2000CC 00 29001F 00	RADELL DI	L •도막교 1 231요
20.00 600.00 $.2076E$ 03 $.2996E$ 00 $.3442E$ 01 $.255$ 20.00 640.00 $.2127E$ 03 $.3043E$ 00 $.3576E$ 01 $.260$ 20.00 660.00 $.2178E$ 03 $.3091E$ 00 $.3576E$ 01 $.260$ 20.00 660.00 $.2230E$ 03 $.3137E$ 00 $.3643E$ 01 $.262$ 20.00 660.00 $.2232E$ 03 $.3137E$ 00 $.3643E$ 01 $.262$ 20.00 660.00 $.2232E$ 03 $.31229E$ 00 $.3777E$ 01 $.268$ 20.00 700.00 $.2332E$ 03 $.3229E$ 00 $.3777E$ 01 $.268$ 20.00 740.00 $.2332E$ 03 $.3229E$ 00 $.3910E$ 01 $.270$ 20.00 740.00 $.2442E$ 03 $.3220E$ 00 $.3910E$ 01 $.270$ 20.00 760.00 $.2442E$ 03 $.3220E$ 00 $.3910E$ 01 $.275$ 20.00 780.00 $.2442E$ 03 $.3454E$ 00 $.4043E$ 01 $.275$ 20.00 780.00 $.266EE$ 03 $.3498E$ 00 $.4177E$ 01 $.283$ 20.00 840.00 $.2775E$ 03 $.3545E$ 00 $.4243E$ 01 $.283$ 20.00 840.00 $.2775E$ 03 $.3576E$ 00 $.4376E$ 01 <td>20.00</td> <td>- <u>500,00</u> 500 40</td> <td>SUSAL AR</td> <td>ачазы ий</td> <td>. 33256 B.</td> <td>L .L</td>	20.00	- <u>500,00</u> 500 40	SUSAL AR	ачазы ий	. 33256 B.	L .L
23.00 623.00 $2127E$ 03 $3043E$ 00 $3539E$ 01 258 23.00 643.00 $2178E$ 03 $3091E$ 00 $3576E$ 01 260 23.00 660.00 $2230E$ 03 $3137E$ 00 $3643E$ 01 262 20.00 660.00 $2232E$ 03 $3137E$ 00 $3643E$ 01 262 20.00 680.00 $2232E$ 03 $3137E$ 00 $3777E$ 01 263 20.00 700.00 $2232E$ 03 $3229E$ 00 $3777E$ 01 263 20.00 700.00 $2333E$ 03 $3275E$ 00 $3943E$ 01 270 20.00 740.00 $2442E$ 03 $3320E$ 00 $3910E$ 01 273 20.00 760.00 $2496E$ 03 $3366E$ 00 $3977E$ 01 275 20.00 760.00 $2551E$ 03 $3454E$ 00 $4110E$ 01 280 20.00 800.00 $2662E$ 03 $3498E$ 00 $4177E$ 01 281 20.00 800.00 $2718E$ 03 $3585E$ 00 $4243E$ 01 283 20.00 860.00 $2775E$ 03 $3585E$ 00 $4376E$ 01 283 20.00 860.00 $2775E$ 03 $3578E$ 00 $4442E$ 01 294 20.00 860.00 <td>20.00</td> <td>630.00</td> <td>.2020E 03</td> <td>2946E 00</td> <td>.3442F 0</td> <td>i tear</td>	20.00	630.00	.2020E 03	2946E 00	.3442F 0	i tear
20.00 643.00 $2178E$ 03 $3091E$ 00 $3576E$ 01 260 20.00 663.00 $2230E$ 03 $3137E$ 00 $3643E$ 01 262 20.00 683.00 $2232E$ 03 $3134E$ 00 $3710E$ 01 265 20.00 700.00 $2233E$ 03 $3229E$ 00 $3710E$ 01 268 20.00 700.00 $2233E$ 03 $3229E$ 00 $3943E$ 01 270 20.00 740.00 $2233E$ 03 $3229E$ 00 $3943E$ 01 270 20.00 740.00 $2442E$ 03 $3320E$ 00 $3943E$ 01 273 20.00 760.00 $2442E$ 03 $3326E$ 00 $3977E$ 01 273 20.00 760.00 $2442E$ 03 $33454E$ 00 $4043E$ 01 278 20.00 780.00 $2551E$ 03 $3449E$ 00 $4110E$ 01 280 20.00 830.00 $2662E$ 03 $3498E$ 00 $4177E$ 01 283 20.00 840.00 $2778E$ 03 $3585E$ 00 $4336E$ 01 283 20.00 860.00 $2778E$ 03 $3585E$ 00 $4376E$ 01 287 20.00 860.00 $2778E$ 03 $3713E$ 00 $4376E$ 01 293 20.00 860.00 </td <td>20.00 21.111</td> <td>626,60</td> <td>.2127F A3</td> <td>30435 00</td> <td>.35895 0</td> <td>.258</td>	20.00 21.111	626,60	.2127F A3	30435 00	.35895 0	.258
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	648.00	.2178F 03	.3091E 00	.3576E 00	260
29.00 680.00 $.2282E$ 03 $.3184E$ 00 $.3710E$ 01 $.265$ 20.00 700.00 $.2335E$ 03 $.3229E$ 00 $.3777E$ 01 $.268$ 20.00 720.00 $.2388E$ 03 $.3275E$ 00 $.3943E$ 01 $.270$ 20.00 740.00 $.2442E$ 03 $.3320E$ 00 $.3910E$ 01 $.273$ 20.00 760.00 $.2442E$ 03 $.3366E$ 00 $.3977E$ 01 $.275$ 20.00 760.00 $.2495E$ 03 $.3366E$ 00 $.3977E$ 01 $.275$ 20.00 760.00 $.2495E$ 03 $.3366E$ 00 $.3977E$ 01 $.275$ 20.00 780.00 $.2551E$ 03 $.3419E$ 00 $.4043E$ 01 $.280$ 20.00 830.00 $.2662E$ 03 $.3498E$ 00 $.4110E$ 01 $.283$ 20.00 840.00 $.2718E$ 03 $.3585E$ 00 $.4243E$ 01 $.283$ 20.00 840.00 $.2775E$ 03 $.3585E$ 00 $.4243E$ 01 $.283$ 20.00 860.00 $.2775E$ 03 $.3585E$ 00 $.4376E$ 01 $.287$ 20.00 860.00 $.2839E$ 03 $.3670E$ 00 $.4442E$ 01 $.293$ 20.00 $.920.00$ $.2947E$ 03 $.3773E$ 00 $.4509E$ 01 <td>29.00</td> <td>660.00</td> <td>2230E 03</td> <td>.3137E 00</td> <td>.3643E 0</td> <td>262</td>	29.00	660.00	2230E 03	.3137E 00	.3643E 0	262
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	20.00	680.09	.2282E 83	.3184E 00	.3710E 0:	.265
20.00 720.00 $.2388E$ 03 $.3275E$ 00 $.3910E$ 01 $.270$ 20.00 740.00 $.2442E$ 03 $.3320E$ 00 $.3910E$ 01 $.275$ 20.00 760.00 $.2496E$ 03 $.3366E$ 00 $.3977E$ 01 $.275$ 20.00 780.00 $.2551E$ 03 $.3416E$ 00 $.4043E$ 01 $.278$ 20.00 800.00 $.2666E$ 03 $.3454E$ 00 $.4110E$ 01 $.280$ 20.00 820.00 $.2662E$ 03 $.3498E$ 00 $.4177E$ 01 $.281$ 20.00 840.00 $.2775E$ 03 $.3585E$ 00 $.4243E$ 01 $.283$ 20.00 840.00 $.2775E$ 03 $.3585E$ 00 $.4243E$ 01 $.283$ 20.00 860.00 $.2775E$ 03 $.3585E$ 00 $.4243E$ 01 $.283$ 20.00 860.00 $.2775E$ 03 $.3585E$ 00 $.4376E$ 01 $.287$ 20.00 860.00 $.2831E$ 03 $.3670E$ 00 $.4442E$ 01 $.293$ 20.00 $.920.00$ $.2947E$ 03 $.3713E$ 00 $.4509E$ 01 $.293$ 20.00 $.920.00$ $.2947E$ 03 $.3775E$ 00 $.4575E$ 01 $.293$ 20.00 $.920.00$ $.3064E$ 03 $.3798E$ 00 $.4778E$ 01 <	20.00	766.68	.23355 03	.3229E 00	.3777E 9:	.268
20.00 740.00 .2442E 03 .3320E 00 .3910E 01 .273 20.00 760.00 .2496E 03 .3366E 00 .3977E 01 .275 20.00 760.00 .2551E 03 .3416E 00 .4043E 01 .278 20.00 800.00 .2666E 03 .3454E 00 .4110E 01 .280 20.00 820.00 .2662E 03 .3454E 00 .4177E 01 .281 20.00 820.00 .2662E 03 .3498E 00 .4177E 01 .281 20.00 840.00 .2718E 03 .3585E 00 .4243E 01 .283 20.00 860.00 .2775E 03 .3585E 00 .4376E 01 .283 20.00 860.00 .2775E 03 .3670E 00 .4442E 01 .290 20.00 920.00 .2947E 03 .3713E 00 .4509E 01 .293	20.00	728.00	.2338E 03	.3275E 00	.3343E 0:	.270
20.00760.00.2496E03.3366E00.3977E01.27520.00780.00.2551E03.3410E00.4043E01.27820.00800.00.2606E03.3454E00.4110E01.28020.00820.00.2662E03.3498E00.4177E01.28120.00820.00.2662E03.3498E00.4177E01.28320.00840.00.2775E03.3585E00.4243E01.28320.00860.00.2775E03.3585E00.4309E01.28520.00860.00.2775E03.3628E00.4376E01.28720.00880.00.2831E03.3670E00.4442E01.28720.00980.86.2889E03.3678E00.4509E01.29320.00920.00.2947E03.3713E60.4509E01.29320.00940.00.3064E03.3796E00.4642E01.29220.00960.00.3183E03.3837E00.4708E01.29420.001000.00.3182E03.3918E00.4774E01.29520.001020.00.3242E03.3918E00.4840E01.29920.001049.00.3202E03.3959E00.4906F	20.00	749.00	.2442E 03	.33295 00	.3910E 0:	.273
20.00 780.00 .2551E 03 .3410E 00 .4043E 01 .278 20.00 800.00 .2606E 03 .3454E 00 .4110E 01 .280 20.00 820.00 .2666E 03 .3458E 00 .4177E 01 .281 20.00 840.00 .2718E 83 .3542E 00 .4243E 01 .283 20.00 860.00 .2775E 03 .3585E 00 .4243E 01 .283 20.00 860.00 .2775E 03 .3585E 00 .4209E 01 .283 20.00 860.00 .2831E 03 .3585E 00 .4376E 01 .287 20.00 960.00 .2831E 03 .3670E 00 .4442E 01 .296 20.00 960.00 .29347E 03 .3713E 00 .4509E 01 .293 20.00 940.00 .3064E 03 .3796E 00 .4642E 01 .294	20.00	768.00	.2496E 03	.3366E 00	.3977E 0:	.275
20.00 800.00 .2606E 03 .3454E 00 .4110E 01 .280 20.00 820.00 .2662E 03 .3498E 00 .4177E 01 .281 20.00 840.00 .2718E 03 .3542E 00 .4243E 01 .283 20.00 860.00 .2775E 03 .3585E 00 .4209E 01 .283 20.00 860.00 .2775E 03 .3585E 00 .4243E 01 .283 20.00 860.00 .2775E 03 .3585E 00 .4376E 01 .285 20.00 880.00 .2831E 03 .3628E 00 .4442E 01 .295 20.00 980.00 .2889E 03 .3713E 60 .4509E 01 .293 20.00 940.00 .2947E 03 .3754E 00 .4575E 01 .293 20.00 940.00 .3064E 03 .3796E 00 .4642E 01 .292	20.00	789.00	.2551E 03	.3410E 00	.4043E 0:	.278
20.00 820.00 .2662E 03 .3498E 00 .4177E 01 .281 20.00 840.00 .2718E 03 .3542E 00 .4243E 01 .283 20.00 860.00 .2775E 03 .3585E 00 .4309E 01 .283 20.00 860.00 .2775E 03 .3585E 00 .4376E 01 .285 20.00 880.00 .2831E 03 .3628E 00 .4376E 01 .287 20.00 980.00 .2831E 03 .3628E 00 .4442E 01 .290 20.00 980.00 .2889E 03 .3670E 00 .4442E 01 .293 20.00 920.00 .2947E 03 .3713E 00 .4509E 01 .293 20.00 940.00 .2005E 03 .3798E 00 .4642E 01 .293 20.00 960.00 .3182E 03 .3837E 00 .4798E 01 .295	ેરે. ઇઇ	899.00	.2606E 03	.3454E 00	4110E 0:	1 .289
20.00 840.00 .2718E 03 .3542E 00 .4243E 01 .283 20.00 860.00 .2775E 03 .3585E 00 .4309E 01 .285 20.00 880.00 .2831E 03 .3628E 00 .4376E 01 .287 20.00 980.00 .2831E 03 .3670E 00 .4442E 01 .290 20.00 980.00 .2839E 03 .3670E 00 .4442E 01 .290 20.00 980.00 .2947E 03 .3713E 00 .4442E 01 .293 20.00 920.00 .2947E 03 .3713E 00 .4509E 01 .293 20.00 940.00 .2005E 03 .3754E 00 .4575E 01 .293 20.00 940.00 .3064E 03 .3795E 00 .4642E 01 .294 20.00 950.00 .3182E 03 .3878E 00 .4774E 01 .295	20.00	820.00	.2662E 03	.3498E 00	.4177E 0:	.281
20.00 360.00 2775E 03 3585E 00 4376E 01 285 20.00 880.00 2831E 03 3628E 00 4376E 01 287 20.00 980.00 2889E 03 3670E 00 4442E 01 290 20.00 980.00 2947E 03 3713E 00 4442E 01 290 20.00 920.00 2947E 03 3713E 00 4509E 01 293 20.00 940.00 2947E 03 3713E 00 4575E 01 293 20.00 940.00 2005E 03 3754E 00 4575E 01 293 20.00 940.00 3064E 03 3798E 00 4642E 01 292 20.00 950.00 3182E 03 3878E 00 4798E 01 295 20.00 1020.00 3242E 03 3918E 00 4840E 01 299 20.00 1040.00	20.00	840.00	.2718E 03	.3542E 00	.4243E 0:	.283
20.00 880.00 .2831E 03 .3628E 00 .4376E 01 .287 20.00 900.00 .2889E 03 .3670E 00 .4442E 01 .290 20.00 920.00 .2947E 03 .3713E 00 .4509E 01 .293 20.00 940.00 .2947E 03 .3713E 00 .4509E 01 .293 20.00 940.00 .2005E 03 .3754E 00 .4575E 01 .293 20.00 940.00 .3064E 03 .3796E 00 .4642E 01 .293 20.00 950.00 .3064E 03 .3796E 00 .4642E 01 .292 20.00 950.00 .3183E 03 .3837E 00 .4774E 01 .295 20.00 1000.00 .3183E 03 .3918E 00 .4840E 01 .295 20.00 1020.00 .3242E 03 .3918E 00 .4906F 01 .299	20.00	860,00	.2775E 03	.3585E 00	.4309E 0:	.285
20.00 988,88 .2889E 03 .3670E 00 .4442E 01 .290 20.00 920.00 .2947E 03 .3713E 00 .4509E 01 .293 20.00 940.00 .2005E 03 .3754E 00 .4575E 01 .293 20.00 940.00 .3064E 03 .3754E 00 .4575E 01 .293 20.00 960.00 .3064E 03 .3798E 00 .4642E 01 .292 20.00 960.00 .3183E 03 .3837E 00 .4708E 01 .294 20.00 1000.00 .3188E 03 .3878E 00 .4774E 01 .295 20.00 1020.00 .3242E 03 .3918E 00 .4840E 01 .299 20.00 1040.00 .3302E 03 .3959E 00 .4906F 01 .299	20.00	889,00	.2831E 03	.3628E 00	.4376E 0	.287
20.00 920.00 .2947E 03 .3713E 00 .4509E 01 .293 20.00 940.00 .0005E 03 .3754E 00 .4575E 01 .293 20.00 940.00 .0005E 03 .3754E 00 .4575E 01 .293 20.00 960.00 .3064E 03 .3795E 00 .4642E 01 .292 20.00 980.00 .3183E 03 .3837E 00 .4708E 01 .294 20.00 1000.00 .3188E 03 .3878E 00 .4774E 01 .295 20.00 1020.00 .3242E 03 .3918E 00 .4840E 01 .299 20.00 1020.00 .3202E 03 .3959E 00 .4906F 01 .299		966,66	.28895 03	.3670E UU	.4442E UI	290
20.00 940.00 .30000 037540 .45750 .293 20.00 960.00 .36640 03 .37960 .46420 .292 20.00 960.00 .36640 03 .37960 .46420 .292 20.00 960.00 .31230 03 .38370 00 .47080 .294 20.00 1000.00 .31820 03 .38780 00 .47740 .295 20.00 1000.00 .31820 03 .39180 00 .48400 .299 20.00 1020.00 .32420 03 .39180 .49460 .299 20.00 1040.00 .33020 .39590 .49460 .49460 .293	20.00	920,00	.2947E M3	.3/13E 80	.45098 01	.293
20.00 200.00 .30641 03 .37951 00 .46421 01 .292 20.00 200.00 .3123E 03 .3837E 00 .4708E 01 .294 20.00 1000.00 .3182E 03 .3878E 00 .4774E 01 .295 20.00 1020.00 .3242E 03 .3918E 00 .4840E 01 .299 20.00 1040.00 .3302E 03 .3959E 00 .4906F 01 .299	28.99 28.00	240,00 020-00	.JUUDE US	.3734E 89 37846 88	.4575E 01	.293
20.00 200.00 .31325 03 .30315 00 .47085 01 .294 20.00 1000.00 .31825 03 .38785 00 .47745 01 .295 20.00 1020.00 .32425 03 .39185 00 .48405 01 .299 20.00 1040.00 .33025 03 .39595 00 .49065 01 .300	្រសារសេស ទំព័រ ម៉ាម	200,99 Saona 604	.08541 83 111000 00	. JY756 80 Decort aa	.4542E U) Arabar 24	.272
20.00 1020.00 .3242E 03 .3918E 00 .4840E 01 .295 20.00 1020.00 .3242E 03 .3918E 00 .4840E 01 .299 20.00 1040.00 .3302E 03 .3959E 00 .490AF 01 .300	2000 - 200 2011 - 1111	TENESSEE	.01505 80 01005 80	-0001E 88 2070E 663	4708E 01 47224E 01	. 2014 1
20.00 1040.00 .3302E 03 .3959E 00 .490AF 61 .259	20.00	1020.00	-0001000 -0100010-000	10010E 00 10410F 00		270 000
	20.00	1040.00	.3302E 03	.0959E NN	.490AF 01	• • • • 7 7

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P	T	H	S (DTILUD OF)	V	СР
(PSIA)	(°⊢)	(BIU/LB)	(BIU/LB-°F)	(FI º / LB) (B	SIU/LB-°F)
					•
	1.10	· • • • • • •	a . * a . ~ a	a shareshire seca	
40.00	60.00	.6464E 01	.1461E-01	.1002E-01	.957
40.00	80.00	.1125E 02	.24656-01	1032E-01	.984
40.00	100.00	.1617E U2	.3440E-01	.1064E-01	1.012
49.00	120.00	.2123E 02	.4389E-01	.1098E-01	1.038
49.00	149.66	.2642E 02	.5316E-01	.11356-01	1.068
40.00	160.00	.3176E 02	.6223E-01	.1174E-01	1.094
40.00	180.00	.3723E 02	.7112E-01	.1217E-01	1.120
40.00	200.00	.42838 02	.7986E-01	.1264E-01	1.150
411.00	220.00	.4858E 02	.8846E-01	.1315E-01	1.178
48.00	240.00	.5447E 02	.9695E-01	.1371E-01	1.208
40.00	260.00	.1303E 03	.2050E 00	.1156E 01	.200
40.00	280.00	.1343E 03	.2104E 00	.1195E 01	203
40.00	399.00	.1384E 03	.2158E 00	.1233E 01	.206
40.00	320.00	.1425E 03	.2211E 00	.1271E 01	.216
40.00	ូ49.00	.1467E 03	.2264E 00	.1308E 01	.214
48.80	369.99	.1509E 03	.2317E 00	.13455 01	.218
40.00	330.00	.1553E 03	.2369E 00	.13825 01	.221
40.00	400.00	.1597E 03	.2422E 00	.1419E 01	.225
48.00	420.00	.1642E 03	.2473E 00	.1456E 01	.227
48.00	440.00	.1687E 03	.2524E 00	.1493E 01	.230
48.00	460.00	.1733E 03	.2574E 00	.1529E 01	.232
40.00	480.00	.1779E 03	.2624E 00	.1565E 01	.235
49.90	500.00	.1827E 03	.2674E 00	.1601E 01	.239
40.00	526.88	.1874E 03	.2724E 00	.1637E 01	.243
46.66	540.00	.1923E 03	.2772E 00	.1673E Ø1	.247
48.88	568.00	.1972E 03	.2821E 00	.1709E 01	.258
48.00	580.00	.2922E 03	.2869E 00	.1745E 01	.253
40.00	600.00	.2072E 03	.2917E 00	.1781E 01	. 255
40.00	620.00	.2123E 03	.2965E 00	.1816E 01	.258
48.00	640.00	.2174E 03	.3012E 00	.1852E 01	.268
40.00	668.00	.2226E 03	.3059E 00	.1888E 01	.262
40.00	680.00	.2279E 03	.3105E 00	.1923E 01	.265
40.00	799,99	.2332E 03	.3151E 00	.1959E 01	.267
40.00	720.00	.2385E 03	.3196E 00	.1994E 01	.270
40.00	740.00	.2439E 03	.3242E 00	.2029E 01	.272
48.00	.60.00	.2493E 03	.3287E UU	.2065E 01	.275
49.66	/80.0U	.2548E 03	.3332E 00	.2100E UI	.278
40.00	890.00	.2603E 03	.3376E UU	2135E 01	.289
40.00	820.00	.2659E 03	.3420E 00	.2170E 01	.283
40.00	840.00	.2715E 03	.3463E UU	.2205E 01	.235
40.00	860.00	.2772E 03	.35078 00	.22402 01	.285
40.00	889.99	.2829E 03	.3550E 00	.2275E 01	.285
49.99	999.99	.2887E 03	.3592E MM	.2310E 01	.288
40.00	920.00	.2944E U3	.3634E ØØ	.2345E 01	.292
40.00	949.00 020 00	.3993E 03	.3676E UU	2381E 01	.292
40.00	760.00	.30626 03	.3718E UU	.2416Ł 01	.293
40.00	980.00	.3121E 83	.3757E 80	.2450£ 01	.294
40.00	1999.99	.3180E 03	.38006 00	.2485E Ø1	.295
40.00	1928.66	.3240E 03	.3840E 00	2520E 01	.299
49.00	1040.00	.33006 03	.3881E 00	.2555E 01	.302

Р	T	H	S (DTH // D. AS)	V	СР
(PSIA)	(°F)	(BIU/LB)	(BIU/LB-°F)	(FT3/LB)	(BTU/LB-°F)
e i a cara	2 13 - 1313	1 A.C. 487 - 174	1 4 5 1 17 - 17 1	1 / / / / D1 ¹¹ / D 1	(74217
1999 - 1999 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997	50.00 Sta daa	1104040 U1 11040 A0	.1401E701 0465E201	10020-01	•231 444
515 - 51 <u>5</u> 617 - 1417	60.00 100 00	12170 00	2400E-01 0440E-01	19366-01	• 204 1 (31:0
	100.00	-1517E 9E 	.34406-01 49006-01	10046-01	1.012
201.00	120.00	-2160E 96 	.43076701 F0126_01	110706-01	1.040
20.00	140.00	-2042C 82 	20010E-01 20000E-01	1130E-01 1174E-01	1.000
200,000 219 809	100.00	-0110E UE 0700E GO	2110E-01	10125-01	1.1024
1951 BE		-01201 DC 	,/112E701 70025201	102AC-01	1.100
00.00 38 00	200.00	RECOL DE ROMAN	.,700E-01 0042E-01	10150-01	1.100
Ele la la	0.00.00 030 00	.4000E NE 64470 A0	.00406-01 02056-01	10100101	1 - 1 CO 1 - 1000
	240.00	2044)E 86 2044)E 86	10500E-01 1050E-00	1.3710-101	1.000
ાન્કાન્કાન્કાન્ડ તોર્કોન્ડ કોર્કોન્ડ	200.00	.58016 85 	.1803E 00 	14336-01 79070-00	1.640
-00-00 18 00	200.00	10048 00 10748 00	2000E 00 2034E 80	- 10712 00 72570 00	. 200 000
213 144	000.00 000.00	1417E 00	01505 00	- 10010 00 70120 888	.200
- C-C D-D- - 213 - 1314	240.00	1417E 03 1420E 00	2000 00 20115 88	-1716E 00 	
200-00 214 1314	240.00	1400E 00 1500E 00		2420F 88	• C 1 D
		1006E 00 1947E 00	2017E 00	9670F 00	
- 1999 - 1999 - 2013 - 1903	380.00 400 00	1991E 00 1991E 00	2079F 88	2422F 00	.EC0 005
28.99	400.00	10710 00 10020 000	24010E 00 2401E 00		1000 - 1000 1000 - 1000
20.00 20 80		1000E 00 1201E 00	24700 AA	Gatal aa	2004
60.00	440.00 260 00	.1001E 00 1700E 00	25228 00	.94100 00 94525 00	.col .coo
		1720E 80 1724E 80	20202 00 05702 AA	.2002E 00 Qoqaf qq	2000 1000
611 614	400,00 500 00		26730 00 26230 00	1012F 01	• ಒಂಬ 2ವಣೆ
- 00.00 - 68 - 69	000.00 500 00	1020E 00 1070E 00	2424F 80	10375 01	• ८ २ ० - ्राज्य
60.00 635 BB	546-66 546-66	10/00 00 19195 03	27225 00	10615 01	• L - O 24 A
607.00 60.00	560 00	19686 03	27715 00	. 10856 01	250
60.00 60 BB	520 GG	201201.	2819F 00	.11086 01	. L
64.44	600.00	20688 03	2868F 99	.1132F 01	.256
6.6.66	620.00	21195 03	.0915F 00	. 1155F 01	259
69.00	640.00	.2170F 03	.2962E 00	.1179E 01	.261
66.00	660.00	.2222PF M3	.30095 00	1202E 01	. 263
644.444	688.00	.2275E M3	.3056E 00	.1225E 01	.266
66.66	299.00	.2328F A3	.3102E 00	.1248E 01	268
60.00	Ten nn	.2381F #3	.3147E 00	.1272E 01	.871
68.89	748.00	.2435E 03	.3193E 00	.1295E 01	273
63.66	760.00	.2489E 03	.3238E 00	.1318E 01	.276
60.00	788.00	.2545E 03	.3283E 00	.1341E 01	.278
69.00	889.00	.2600E 03	.3327E 00	.1364E 01	.280
69.69	820.00	.2656E 03	.3371E 00	.1387E 01	283
68.98	840.00	.2713E 03	.3415E 00	.1409E 01	.285
60.00	860.00	.2770E 03	.3458E 00	.1432E 01	.286
60.00	880.00	.2827E 03	.3501E 00	.1455E 01	286
60.00	900.00	.2884E 03	.35448 00	.1478E 01	288
60.00	920.60	.2942E 03	.35868 00	.1501E 01	290
69.00	940.00	.3000E 03	.3628E 00	.1523E 01	.292
69.99	960.00	.3059E 03	.3670E 00	.1546E 01	.295
60.00	980.00	.3118E 03	.3711E 00	.1569E 01	.296
68.00	1999.90	.3178E 03	.3752E 00	.15912 01	.296
60.00	1020.00	.3238E 03	.3792E 00	.1614E 01	.299
60.00	1040.00	.3298E 03	.38338 00	.1637E 01	.301

·P	т	Н	S	V	СР
(PSIA)	(°F)	(BTU/LB)	(BTU/LB-°F)	(FT³/LB)	(BTU/LB-°F)
88.69	69.00	.6464E 81	.1461E-01	.10028-01	. 957
ភូមិ មិប	20.00	.1125E U2	.24658-01	.1032E-01	.984
86.00	199.69	.1617E 02	.3440E-01	.1064E-01	1.012
្លារ ព្រ	120.00	.2123E 02	.4389E-01	.1098E-01	1.038
Elf, Litt	140.00	.2642E 02	.5316E-01	.1135E-01	1.968
ះថ្ងៃ អូម	169.00	.3176E 02	.6223E-01	.1174E-01	1.094
86.66	189.00	.3723E 02	.7112E-01	.1217E-01	1.120
ូម, អូម	200.00	.4283E 02	.79866-01	.1264E-01	1.155
89.69	229.99	.4858E 02	.8846E-01	.1315E-01	1.178
90.99	240.00	.5447E U2	.96958-01	.1371E-61	1.208
ូម.មួម	260.00	.6051E 02	.1053E 80	.1433E-01	1.249
80.00	280.00	.6671E UZ	.1137E 00	.1583E-01	1.276
86.00	390.00	.1366E US	.2065E 00	.5777E 00	.209
86.00	350.00	.1409E 03	.2119E 80	.5987E UK	.213
80.00	346.69	.1452E U3	.2173E 80	.6195E 08	.218
ូល.ប្រ	360.00	.1495E U3	.2227E 80	.6402E 00	.223
59. DE	389.99	.15406 03	.2280E 00	.66046.06	1 .224
80.80	400.00	.1584E 08	.23335 90	LANDEL NE	ు చెద్ది
80.00	420.00	.16395 93	.2385E 00	. PUSE UE	1. <u></u>
39.99	445.00	.16/58 83	.24375 88	. ZUUL UN	1 •జావాయ సంఘంత
<u>19</u> .499	465.85	.1722E 03	.2487E 00 orone wa	-1373E 86 175001 60	1 .234
201.20	489.00	.1757E 83 Kot76 An	.20386.00	.73871 08 171701 AU) .230 N NAC
50.00	000.00	.1817E 83 	.20088 88	1.111111111111111111111111111111111111) .ピートロ 2 つっぽ
30.00	520.00	.1860E 03 .014E 03	.26376 99 	.(2:0E 05 0:500 00) .CHO >
201-199 10 1014	040.00 623 00	.1714E UO 	STARE OF) .CH(2
00.00 200.00	<u>- 950.00</u> . 500.00	.1750E 80 	.2705E 80 0705E 80		1 1.CUU 1 1.CUU
192 . 92	200.UD 2000.UD	-2010C 00 0020E 00	2000E 00 0000E 88	- 0004E 89) .EU4 N 050
	600.60 203 BB	2003E 03 0115E 60	2000E 00 2001E 00	2010IC 00 20075 00	ು ಕಾದಲಾ ಇಲ್ಲಿ ಭಾಷೆಗೆ
	620.00 620 00	STARF DR	SASAR OD) .LOI 1 263
20.00	660 00	2000 00 2018F 63	29256 00	9976F 80) .200 N 266
98.89	1680.00	.2271F 93	. BHPPE HD	.9459F 00	· .268
84.66	780.00	.2024F FR3	CRASHE BR	9642F NC	1 .271
સર્ચ છેલ	720.00	.9378E 03	.3114E 00	.98258 00	.273
39.39	740.00	.2432E 03	.3160E 00	.1001E 01	.275
80.00	760.00	.2486E 03	.3295E 00	.1019E 01	.278
98.00	785.00	.2542E 03	.32505 00	.1037E 01	.279
80.00	899.99	.2597E 03	.3294E 00	.1055E 01	.280
80.UU	829.99	.26538 03	.33385 00	.10735 01	.283
80.99	848.99	.2710E 03	.33825 00	.10915 01	.285
89.9J	868.00	.2767E 03	.3425E 00	.1109E 01	.286
80.00	889.09	.2824E 03	.34688 00	.11276 01	.288
88.00	999.99	.2882E 03	.35115 00	.1145E 01	.289
80.00	920.00	.2939E 03	.3553E 00	.1163E 01	.290
સ્ત્ર નહે	940.00	.2998E 03	.3595E 00	.1181E 01	.293
80.00	960.00	.3056E 03	.3637E 00	.1199E 01	.295
50.00	980.00	.3:16E 03	.3678E 00	.1216E 01	.297
80.00	1999.99	.3175E 03	.37195 00	.1234E 01	.298
30.00	1020.00	.3235E 03	.37608 00	.12528 01	.300
89.00	1949.09	.3295E 03	.3300E 00	.1276E 01	. 333

Р	т	н	S	v	CP .
(PSIA)	(°F)	(BTU/LB)	(BTU/LB-°F)	(FT 3/LB)	(BTU/LB-°F)
199.00	60.00	.6464E 01	.1461E-01	.1002E-01	.957
199.60	80.00	.1125E 02	.2465E-01	.10325-01	.984
100.00	100.00	.1617E 02	.3440E-01	.1064E-01	1.012
100.00	120.00	.2123E 02	.4389E-01	.1098E-01	1.038
100.00	140.00	.2642E 02	.5316E-01	.1135E-01	1.068
180.00	160.00	.3176E 02	.6223E-01	.1174E-01	1.094
199.09	180.00	.37238 02	.7112E-01	.1217E-01	1.120
100.00	200.00	.4283E 02	.7986E-01	.1264E-01	1.150
199.99	220.00	.4858E 02	.3846E-01	.1315E-01	1.178
100.00	240.00	.5447E 02	.9695E-01	.1371E-01	1.208
188.68	268.90	.6051E 02	.1053E 00	.1433E-01	1.240
160.00	280.00	.6671E 02	.1137E 00	.1503E-01	1.276
100.00	300.00	.7309E 02	.1220E 00	.15826-01	1.312
166.66	329.99	.1398E 03	.2084E UU	.40,0E 00	.225
199.99	340.00	.1443E 03	.3137E UU	.4225E UU	.225
100.00	360.00	.1487E 03	.2191E UU	4391E 00	.215
199.99	386.00	.1533E 03	.2243E 00	.4541E 00	,220
199.99	400.00	.1578E 03	.2296E UU	.4690E UU	.225
199.90	420.00	.1623E 03.	.2348E UU	.4840E UU	.230
199.99	440.00	.1669E V3	.2401E 00	.4990E 00	.235
189.99	469.00	.1716E 63	.2452E UU	.5135E UU	.235
100.00	480.00	.1764E U3	.2503E 00	5280E UU	.235
100.00	500.00	.1812E 03	.2553E 00	.54248 00	.246
166.66	520.00	1860E US	.2604E 00	.5568E UU	.245
<u>199</u> .66	540.00	.1989E 83	.2653E 00	.5708E UU	.247
100.00	560.00	.19598 83	.2702E 00	.58488 88	1.250
100.00	580.00	.500AF 03	.2750E 00	.59868 80	.255
199.00	600.00	.2059E 03	.27995 00	.5124E 09	1.260
100.00	020.00	.2110E 03	.20478 00	.0661E 00 200005 00) .డోపిన సంజరా
100.00	540.00 770 00	.cibet 03	.2870 <u>t</u> 00	.5378E 85 /5001 00	
155.60	500.00 200 00	- CC14C 00 - TOCIAC 00	.2741E 00 0000E 00	.0006E 06 2227E 00) .COO } .070
198.00	1000-00 1766 66		.27000 00 00041 00	2001E 00 2001E 00) .C(U) () .C(U)
100.00	100.00 708 88	.COCUL UO 00748 000	.30040 00 00040 00	2905E 00 2905E 00	・ローン 4 - つうに
100.00	246.00	0400E 90	2126F 88	2020E 00 2020E 00) .C.(0)) .227
100.00	768 88	23900 BC	2172F 00	7205F 00	280 1 280
100.00	220 00	00 100713. 281 192290	2216F 00	.73365 00	1 280
199.99	800.00	2594F 83	.3261F MM	.7468E AF	1 .280
146.56	829.00	.2650E 03	.3305F 00	.7598F AV	1 .283
139.60	840.00	.2707E 03	.3349F 00	.7728E WV	. 285
166.60	860.00	.2764E M3	.3392E 00	7858E 00	. 287
100.00	889.00	.2821E 83	.3435E 00	.7987E 00) .290
169.00	900.00	.2879E 03	.3478E 00	.8119E 00	965.
100.00	920.00	.2937E 03	.3521E 00	.8251E 00	.290
180.00	940.00	.2995E 03	.3562E 00	.83818 00) .293
100.00	960.00	.3054E 03	.3604E 00	.8511E 00) .295
188.00	980.00	.3113E 03	.3645E 00	.8637E 00	.298
199.00	1999.09	.3173E 03	.3687E 00	.8764E 00	.390
199.90	1929.00	.323 3E 03	.3727E 00	.8895E 00	.302
160.00	1040.00	.32938 03	.3768E 00	.9025E NP	. 305

	SUPERHEATED VAPOR						
	P (PSIA)	T (°F)	H (BTU/LB)	S (BTU/LB-°F)	V (FT³/LB) (E	CP STU/LB-°F)	
	339.09	69.99	.6464E 01	.1461E-01	.10020-01	.957	
•	399.00	88.94	1125E 02	.2465E-01	.10326-01	. 984	
	ិភិមិ, មិមិ	166.66	1617E M2	.3449E-01	.1064E-01	1.012	
	ំពុំស. មុំស	128.66	.2123E 92	4389E-01	.1098E-01	1.038	
	ີ່ 255.65	140.00	.2642E 02	.5316E-01	.11350-01	1.068	
	399.99	160.00	.3176E 02	.62232-01	.1174E-01	1.094	
	366.66	180.00	3723E 02	.7112E-01	.1217E-01	1.120	
	399.99	200.00	4283E 02	.7986E-01	1264E-01	1.150	
	399.99	220.00	4858E 02	.8846E-01	.1315E-01	1.178	
	366.66	240.00	.5447E 02	.9695E-01	.1371E-01	1.208	
	399.99	260.00	.6051E 02	.1053E 00	.1433E-01	1.240	
	000.0g	280.00	.6671E 02	.1137E 00	.1503E-01	1.276	
	399.00	399.09	.73898 02	.1220E 00	.15826-01	1.312	
	3130.694	329.99	.7965E 02	.1302E 00	.1673E-01	1.358	
	ៈអេច.អេច	340.00	.8644E 02	.1386E 00	.1780E-01	1.414	
	ូល្ម សម	368.69	.9351E 02	.1470E 00	.19988-01	f : 478	
•	389.00	380.00	.1009E 03	.1556E 00	.20676-01	1.600	
	300.00	400.00	.1089E 03	.1647E 00	.2272E-01	1.786	
	399.90	420.00	.1178E 03	.1745E 00	.25638-01	2.200	
	368.80	440.00	.1587E 03	.2208E 00	.11918 00	.275	
	<u> </u>	460.00	.1641E 03	.2267E 00	.1286E 66	.270	
	395.99	480.00	.1695E 03	.2326E 00	.1369E 00	.265	
	395.09	599.89	.1748E 03	.2381E 00	.1444E 00	.262	
	399.99	520.00	.1801E 03	.2436E 00	.15195 00	.260	
•	389.99	549.00	.1854E 03	.2488E 00	.1586E 00	.262	
	300.00	568.89	.1906E 03	.2541E 00	.1653E 00	.265	
	ા ાલગાનન	580.00	.1959E 83	.2592E 00	.1715E 00	.265	
	360,60	606.00	.2012E 03	.2643E 00	.1778E 00	.265	
•	: 369.99	620.00	.2065E 03	.2692E 00	.1837E 00	.268	
	369.09	640.00	.2119E 03	.2742E 00	.1896E 00	.270	
•	399.99	660.00	.2173E 03	.2790E 00	.1952E 00	.273	
:	CUU,UU	680.09	.2228E 03	.2839E 00	.2008E 00	.275	
	369.66	700.00	.2283E 03	.2886E 00	.2063E 00	.275	
	300.00	729.00	.2338E 03	.2934E 00	.2117E 00	.275	
	389.99	749.09	.2393E 03	.2980E 00	.2170E 00	.280	
	399.99	768.00	.2449E 03	.3027E 00	.2223E 00	.285	
	335.99	789.99	.2506E 03	.3072E 00	.2274E 00	.285	
	300.00	899.09	.2563E 03	.3118E 00	.2326E 00	.285	
	355.99	829.99	.2620E 03	.3163E 00	.2376E 00	.297	
	338.39	840.00	.2677E 03	.3298E 00	.2427E 80	.298	
	369.69	868.89	.2735E 03	.3252E 00	.2476E 00	.292	
	399.99	880.00	.2793E 03	.3296E 00	.2525E 00	.295	
	300.00	900.00	.2852E 03	.33395 00	.2574E 00	· 2'9'	
	399.99	920.00	.2911E 03	.3352E 00	.2623E 00	.295	
	3515.55	945.66	.2971E 83	3424E 00	.26/1E 00	.295	
	399.99 San os	90 0.0 0	.GUGUE UG	.3467E UU	.2719E 00	.295	
	-31363 - 6163 	939.69	.JUBUE US	.SOUGE UN	.2766E UU	.297	
	300.00 1990 - 445	1000.00	.JIDUE UB	.3001E 80	.2814E UU	.388	
	000.00	1020.00	.3211E US	.SOURCE UN	.2861E UU	.365	
	996 .9 0	1040.00	. <i>361</i> 16 US	- 633E 90	.29035 00	.395	

110

P (PSIA)	T (°F)	H (BTU/LB)	S (BTU/LB-°F)	· V ·(FT³/LB)	CP (BTU/LB-°F)
		11 A 11 AV ⁴ - 12 B	6 .4.5 6 177 . 174 6	1 (2) 2 (117) (2) 4	
i de la constante de la consta La constante de la constante de		- 0404K - 11	.14016-01 	· 16966-191	. •207
uituta kata. Minana nana	لااية ولاتتر	.11206 02	.24601-01	.16326-01	
	100.00	- 1517E 86	.34496-01	.18646-01	. 1.912
000.00	129.00	.21230 92	4389E-01	.1098E-01	1.933
000.00	146.66	2642E UZ	.0316E-01	.1135t-43	1.868
0.00.00	166.66	3176E UE	.62236-01	.11/4E-01	1.594
000.00	180.00	.3.23E 02	. 112L-01	.1817E-41	1.120
509.00	500.00	.4283E 92	./986E-01	.1264L-01	. 1.155
500.00	550.00	.4858E 02	.8846E-01	.13158-01	. 1.172
560.00	240.00	.5447E 02	.96958-01	.1371E-01	1.200
589.00	260.00	.60515 02	.1853E 88	.1433E-01	1.248
588.88	280.00	.6671E 82	.1137E 00	.1503E-01	1.276
566.66	380.89	.7309E 62	.1220E 00	.1582E-01	1.012
599.99	320.80	.79655 02	.1302E 00	.16735-01	1.358
566.66	346.66	.86448 92	.1386E 00	.1789E-01	1.414
566.66	360.00	.9351E 82	.1470E 00	.1908E-01	1.478
580.80	389.99	.1009E 00	.1556E 00	.2067E-01	1.600
200.00	489.88	.1089E 03	.1647E 00	.2272E-01	1.780
599.00	428.80	.1178E-03	.1/45E UU	.25636-01	2.200
539.00	446.85	.1288E 03	.1862E 60	.30771-01	2.643
500.00	465.55	1445E 49	8296£ 48	- 4006E 45	***
599.99	488.00	86/2E 49	49/8E 49	2484E 46	****
000.00	500.00	,1698£ 93	1199E 88	.58748-01	. 425
000.00	529.99	.17096.03	.2295E 00	.62465-01	.345
	040.00 C/0 664	.i/bbt US	.dobit 00	. (8955-59) 17195-66	l .320 Doge
	UDELEE Konie enge	.10000 80	204678 66 194046 66	PERSONAL TRANSPORT	UUU
500.00	200.00	ALOZEE UN Antiste un	.2404E 80 officiencia	- 0600 (E.778)) 06000787 - 64	L .C70 □
589.00	680.00 200 00	.1706E 80 004087 00	-COMEE 00 	.00748703 00706-01	L .278
Hader ein	020.00 2413 003	- 19195 60 	.2070E 00 0250E 00	,7017ET01 0020FLA1	L .CO/ L .and
	640.00	.1900E 93 0102E 60	2000E 00 2001E 00	.2002E701 13739E 30	L PEQUE N Sener
HEELING	698.00	0400L 00	OPEOR DO	1920E 00 1920E 00	2 .COU R : ::::::::::::::::::::::::::::::::::
- 1960 AG	200 . 00	22332 32	22011 AA	11100 OC 00	2 - EGU R - GO7
588.89	220.00 220 66	22475 82	2951E 00	115aF 00	2 .LO; 1 003
589.94	740.00	209998 60	220010 00 2299F 00	11926 00	a (2026) A (2026)
589.09	769.09	.2413E 03	.2947F 86	. 1230E OK	1 2444
566.66	780.00	.F471F M3	.2994F 00	. 1266E BE	· · · · · · · · · · · · · · · · · · ·
569.00	886.86	LASAAL NG	.3941E 00	. 10MPF Du	ն հերչը։ Դերնել
555.00	828.66	.2588E 03	.3987E 00	.1336E AM	-
566.00	840.00	.2646E 03	.3133E 00	.1871F M	- 1995 -
566.00	868.98	.2706E 83	.0177E 00	1405E 00	3 .295
588.00	880.00	.2765E U3	.32222E 00	.1438E 00	295
589,00	900.00	.23246 03	.3266E 90	.1471E 00	297
500.00	928.60	.2884E 03	.3310E 00	.1504E 00	300
565.68	940.00	.29448 03	.33538 00	.1535E 00	300
500.00	960.00	.30055 03	.3396E 00	.15678 00	.300
500.00	980.00	.3066E U3	.3439E 00	.1599E 00	.385
599.00	1000.00	.3126E 03		.16305 00	.319
509.00	1020.00	.3188E 93	.3522E 00	.1660E 00	.310
588.00	1040.00	.3249E 03	.3564F 00	. 1691F 00	1 210

P (PSIA)	T (°F)	H (BTU/LB)	S (BT11/LB-°F)	V (FT 3 /I B	
(1011)		(2,2)20)			
				-	
24141.4144	111.1111	.64646 01	. 1461F-01	1645F-1	ai .957
ិទាម ឆម	88.40	.1125E 02	.2465E-01	.1032E-	31 .984
788.99	เย่น.ย่น	.1617E 02	.3443E-01	.1064E-	01 1.012
799.99	136.00	2123E 02	.4389E-01	1998E-I	31 1.038
789.99	140.00	.2642E 62	.5316E-01	.1135E-	31 1.068
700.09	160.00	.0176E 02	.6223E-01	.1174E-0	31 1.0 94 '
789.00	180.00	.37238 02	.7112E-01	.1217E-	01 1.120
700.00	200.00	.4283E 02	.7986E-01	.1264E-	01 1.150
789.99	220.00	.4858E 02	.8846E-01	.1315E-	01 1.178
700.00	240.00	.5447E 02	.9695E-01	.13715-0	01 1.208
788.68	269.99	.6051E 02	.1053E 00	.1433E-	01 1.240
766.96	289.00	.6671E 02	.1137E 00	.1503E-	01 1.276
700.00	300.00	.7309E 02	.1220E 00	.15828-	01 1.312
760.00	029 . 90	.7965E 02	.1382E 00	.1673E-	61 1.358
768.00	343.00	.8644E 02	.1386E 00	.1780E-	01 1.414
766.66	360.00	.9351E 82	.1470E 00	.1908E-	01 1.478
700.00	388.00	.1009E 03	.1556E 00	.20676-	01 1.600
700.00	400.00	.1889E 83	.1647E 00	.22725-	01 1.780
788.00	420.00	.1178E 03	.1745E 00	.25638-	01 2.200
799.00	440.00	.1288E 03	.1862E 00	.3877E-	01 2.640
768.99	460.00	1445E 49	8296E 48	4006E	45#######
708.00	480.00	8672E 49	4978E 49	24048	46#######
785.66	500.00	1590E 50	9126E 49	4407E	46#######
788.99	520.00	2313E 50	1327E 50	6410E	46#######
788.89	540.00	3035E 50	1742E 50	8413E	46#######
200.00	560.00	.1768E 03	.2349E 00	.5844E-	01 .313
280.00	580.00	.1831E 03	.2409E 00	.6263E	.310
799.99	600.00	.1893E 03	.2468E 00	.6682E-	01 . 386
700.00	628.88	.1954E 03	.2525E 00	.7061E-	01 .305
700.00	640.00	.2015E 03	.2582E 00	.7440E-	91 . 303
200.00	660,00	.2876E 03	.2636E UU	.7798E-	01 .301
<u> </u>	689.00	.2136E 03	.2690E UU	.81556-	01 .299 J
	199.99	.2196E 03	.2741E 00	.84936-	01 .2 99
700.00	120.00	.2206E 03	.E793E 00	.88305-	91 .399 At 1999
200.00	140.00	.calbt 83	.2343E 80	. 9101E-	91 .399 At 1082
100.00	150.00	.236E 93	.2073E 80% 		91 - J99 G1 - J964
1999.999 17:49 - 669	100.00 (0.0 00	-2435E 90 	.C7410 88 000000 800	. 7777CT 100000	91 .091 MG 1494
1995-885 1296-685	000.00 000 AGA	-1470L 90 Verze 140	20220E 00 20227E 00	. 1990E 1997E	00 .301 00 .301
200.00	960.00 Ván Ak	- LUUDE DU VIIIE AV		1001C	90 .001 1919 - 1919 -
i consentational Constante sinte		.2010E 00 Sayar 60	2129FC 88	10010	99 .091 1313 13130
700.00		.00707E 80		.1070C 11000	20 .2000 1313 131313
700.00 708 88	488.88	2799F 83	2228F 80	11516	00 .000 1313 - 1313
	actioned Action	PREADE AR	2245F 00	11205	00 .004 NA 204
288.00	-20,00 446 86	,2920E 83	.3393F 99	12055	00 .000 88 285
788 BB	960.00	.2981F 83	.3352F AA	.1231F	ос .000 Ий .204
76161 6161	980.00	.3043F 63	.3395F AA	. 12575	 AA .384
799.99	1999 . 99	.3104F 03	.3438F AA	12835	ий "R14
200.00	1828 86	.3166F 03	.348AF AA	1399F	йй <u>.</u> 314
766.00	1040.00	.32295 03	3522E ØØ	.1334F	йй .314

P (PSIA)	۲ (°F)	H (BTŲ/LB)	S (BTU/LB-°F)	V (FT³/LB)	CP (BTU/LB-°F)
966.0d	មម,ស្រ	.6464E 01	.1461E-01	.10028-01	الا المالية 1912 - المالية
990.03	89.00	.1125E 02	.2465E-01	.10325-01	. 4934
-30.00	, មម, មម	.1617E 02	.3449E-01	.1964E-01	1.912
111. 111	120.00	.2123E 02	,4389E−01	.10930-01	1.038
មល់ច, ចូម	140.00	.2642E U2	.5316E-01	.1135E-01	1.068
909,00	160.00	.2176E 02	.6223E-01	.1174E-01	1.094
900.00	180.00	.07238 02	.7112E-01	.12178-01	1.129
999,00	200.00	.4283E U2	.7986E-01	.1264E-01	1.150
900.00	225.05	.4858E 02	.0846E-01	.1315E-91	L 1.178
900.00	240.00	.5447E 62	,9695E-01	,13715-01	1.208
988.89	268.00	.6051E 82	.1053E 00	.1433E-01	1.248
999.09	288.00	.6671E 02	.1137E 00	.1503E-01	1.276
900.00	388.00	.73995 92	.1220E 00	.15826-01	1 1.318
989.89	320.00	.79658 02	.1382E 00	.16735-03	1 1.358
ំ១២.មហ	348.96	.8644E 02	.1386E 00	.17805-01	1 1.414
988.88	368.00	.9351E 02	.1470E 00	.19988-00	1 1.478
998.00	389.00	.1009E 03	.1556E 00	.2067E-0	1 1.600
990.00	460.68	.1889E 83	.1647E 00	.2272E-0	1 1.789
999.00	428.00	.11785 03	.1745E 00	.2563E-0	1 2.200
988.00	449.00	1388E 03	,1862E 00	.3077E-0	1 2.640
900.00	ុំស្រុំ អូម	1445E 49	8296E 48	4006E 45	5########
988.08	438.00	8672E 49	4978E 49	2404E 40	5#######
906.00	500.00	-:1590E 50	9126E 49	4407E 40	5########
990.00	529.09	2313E 50	1327E 50	6410E 4/	5#######
900.00	540.00	-13035E 50	1742E 50	8413E 4	5########
990.00	560.00	3758E 50	2157E 50	1042E 4	2########
986.00	580.00	4481E 50	25728 50	1242E 4	2########
966.66	600.00	52838 50	29878 50	1442E 4	?########
999.69	629,00	.1898E 03	.2454E 00	.4743E-0	1322
900.00	640.00	.1962E U3	.2513E 00	5019E-0	1 .021
900.00	660,00	.2026E 03	.2570E 00	.52988-0	1 .317
995.55	689.99	.2089E 03	.26278 00	.5577E-0	1 .313
389.00	789.89	.2152E 03	.2681E 00	.58508-0	1.311
999.09	720.00	.2215E 03	.2735E 60	.6123E-0	1.310
900.00	740.00	.2277E 03	.2787E 00	.63858-0	1.310
999.99	760.00	.23395 03	.2839E 00	.6648E-0	1.310
999.09	783.00	.2401E 03	.2889E 00	.6897E-0	1 .369
999.99	800.00	.2463E 03	.29398 00	.71468-0	1 .337
986.00	820.00	.2524E 03	.2987E 00	.73858-0	1.397
399.00	849.06	.2586E 63	.3035E 00	.76235-0	1 .007
900.00	360.80	.26485 03	.3082E 80	78526-0	1 .389
966.00	885.05	.27995 03	.3128E 00	.8081E-0	1 .311
ាមាច ជាម	966.66	.2771E 03	.3174E 00	.8303E-0	1 .312
985.88	920.00	.28335 03	.3220E 00	.8524E-0	1 .312
900.00	949.00	.2895E 03	.3264E 00	.8739E-0	1.318
999.09	966.08	.2958E 03	.3308E 00	.8953E-0	1.308
્યતઘર, દાહ	980.00	.3920E 03	.3352E 00	.9168E-0	1 .313
200.00	1999.00	.39828 03	.33955 00	.9371E-0	.318
999.00	1929.00	.0145E 03	.3438E 00	.9574E-0	1 .318
909,99	1040.00	.3208E 03	.3480E 00	.9777E-0.	1 .218

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P (PSIA)	T (°F)	H (BTU/LB)	S (BTU/LB-°F)	V (FT ³ /IB)	
		(=	(=:::::::::;	(11 /20)	
1169.69	66.00	.6464E 81	.1461E-01	.1002E-01	.957
1180.00	80.00	.1125E 02	.2465E-01	.1832E01	. 984
1166.00	100.00	.1617E 02	.3440E-01	.1064E-01	1.012
1100.00	120.00	.2123E U2	.4389E-01	.1098E-01	1.038
1100.00	140.00	.2642E 02	.5316E-01	.1135E-01	1.068
1100.00	160.00	.3176E 02	.6223E-01	.1174E-01	1.094
1100.00	180.00	.3723E 02	.7112E-01	.1217E-01	1.120
1100.00	200.00	.4283E 02	.7986E-01	.1264E-01	1.150
1169.00	220.00	.4858E 02	.8846E-01	.1315E-01	1.178
1166.60	240,00	.5447E 02	.9695E-01	.1371E-01	1.208
1100.00	260.00	.6051E 02	.1653E 00	.1433E-01	1.240
1196.96	280.00	.6671E 02	.1137E 00	.1503E-01	1.276
1100.00	366.66	.7309E 02	.1220E 00	.1582E-01	1.312
1100.00	320.00	.7965E 02	.1302E 00	.1673E-01	i. 358
1100.00	340,90	.8644E 02	.1386E 00	.1780E-01	1.414
1100.00	360.00	.9351E 02	.1470E 00	.1903E-01	1.478
1100.00	388.99	.1009E 03	.1556E 00	.2067E-01	1.600
1160.00	400.00	.1089E 03	.1647E 00	.2272E-01	1.780
1100.00	420.00	.1178E 03	.1745E 00	.2563E-01	5.200
1190.00	440.00	.1288E 03	.1862E 00	.3077E-01	2.640
1166.60	468.80	1445E 49	8296E 48	4006E 45)########
1100.00	488.00	8672E 49	4978E 49	2404E 46	****
1144.00	500.00	1590E 50	9126E 49	4497E 46	**** ****
1100.00	520.00	2313£ 50	1327E 50		***** ***
1166.00	540.00	-,3835E 58	1/42E 50	8413E 46	*****
1100.00	560.00	3758E 50	2107E 30	1042E 47	**********
1100.00	380.00	44816 00	2072E 00		**********
1100.00	500.00	02031 00 20025 80	C70(E 00 	- THIHEE HI 12400 40	*********
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1100.00	840.00 223 33	10047E 30 1008E 80	2522F 44	20905 HI 20055-01	пппппп (200
1100.00	650.00	.1779E 03 0655E 00	2521F 66	40292F-01	
	200.00	2000E 00 2119E 02	26345 66	4254F-01	012
1100.00	720.00 720 00	21228 63	.26901 00	.4469F-01	
1100.00	240.00	.2246F 03	.2743F MA	. 46808-01	.315
198.99	760.00	.23995 83	.2796E MB	4891F-01	.315
1100.00	780.00	.2372E 83	2847E 00	.5094E-01	.313
1198.00	800.00	.2435E 03	.2897E 00	.5296E-01	.311
1100.00	820.00	.2498E 03	.2946E 00	.5491E-01	.311
1103.00	849.80	.2560E 03	.2996E 00	.5686E-01	.311
1100.00	860.00	.2623E 03	.3043E 00	.5873E-01	.314
1100.00	880.00	.2685E 03	.3090E 00	.6060E-01	.316
1100.00	900.00	.2748E 03	.3136E 00	.62428-01	.316
1100.00	920.00	.2810E 03	.3183E 00	.64238-01	.316
1199.00	940.00	.2874E 03	.3227E 00	.6599E-01	.314
1100.00	960.00	.2937E 03	.3272E 00	.6774E-01	.312
1100.00	980.00	.3000E 03	.3316E 00	.6945E-01	.316
1199.99	1999.99	.3962E 03	.3360E 00	.7117E-01	.326
1100.00	1929.00	.31268 03	.3403E 00	.72826-01	.328
1100.00	1040.00	.3190E 03	.3446E 00	.7448E-01	.321

P (PSIA)	Т (°F)	H (BTU/LB)	S (BTU/LB-°F)	V. (FT³/LB)	CP (BTU/LB-°F)
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a a des ees. A fais a siste	500.000 San 140	READER OF AND	2465F-01	111226-01	
1200.00	Sector Free National Anna	1212UD UD 1217D 000	24406-01	10641-01	1.012
	199.00	A COLLEGE AND A	43396-01	10985-01	1.638
1.26161-640	140.00	12.4.217 (1.2	.5316E-01	1135E-0	1.068
1 Roten Eiten	140.00	ATER OF	6223E-01	.1174E-03	1.1944
1.5.36.60	1991.00	. 37236 02	.7112E-01	.1217E-0	1.120
1344,49	200.00	.4283E M2	.7936E-01	.1264E-0	1.158
1000.00	220,03	4858E 02	.8846E-01	.1315E-01	1 1.178
1366.60	240.00	.5447E 02	.969SE-01	.1371E-0:	1.208
1366.66	260.60	.6051E 82	.1853E 99	.1433E-0	1 1.240
1030.00	289.69	.6671E 02	.1137E 00	.1503E-0	1 1.276
1300.00	366.66	.7309E 02	.1220E 00	.15828-0	1 1.312
1300.00	320.05	.79655 02	.1302E 00	.1673E-0	1 1.358
1339.00	្លា4មិ.មេច	.8644E 02	.1386E 00	.1780E-0	1 1.414
1300.00	360.00	.9351E 02	.1479E 00	.19885-0	1 1.478
1300.00	380.00	.1009E 03	.1556E 00	.20678-0	1 1.600
1300.00	400.00	.1889E 83	.1647E 00	.2272E-0	1 1.788
1300.00	420.00	11178E 03	.1745E 00	.25638-0	1 2.200
1300.00	440.00	.1288E 03	.1862E 00	.3077E-0	1 2.649
1300.00	460.00	1445E 49	8296E 48	4006E 4	5########
1399.99	480.00	8672E 49	4978E 49	2404E 4	5########
1399.99	500.00	1590E 50	9126E 49	4407E 4	5########
1386.00	520.00	2313E 50	1327E 50	6410E 4	5***
1300.00	540.00	30358 50	-,1742E 50	8413E 4	6#######
1380.00	569.99	4.3758E 50	2157E 59	1042E 4	~****
1300.00	580.00	4481E 50	25/2E 50	-1242E 4	*******
1300.00	600.00	5203E 50	29878 50	1442E 4	*******
1300.00	620.00	5926E 50	3401E 50	1643E 4	
1300.00	640.00	6649E 50	38166 30 20015 60		~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~
1300.00	555.00	7371E 30	423IE 00 42425 50	- T.E040E 4 	*********
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1200.00	786.66	SRAAF 6R	28155 00	4367F-B	1 .014
1.300.00	200.00	-2412F 03	.2866E 89	4531E-0	1 .313
1366.66	829.96	.2476E 03	.2915E 00	4693E-0	1 .313
1309.00	840,00	.2539E H3	2965E 80	.4854E-0	1 .313
1335.69	869.95	.2691E 93	.3013E 00	.5011E-0	1 .316
1399.99	880.00	.2664E 03	.3961E 00	.5169E-0	1 .318
1390.00	999,99	.2728E 03	.3107E 00	.5323E-0	1 .818
1330.00	920.00	.2791E 03	.3154E 00	.5477E-0	i .318
1390.00	940.00	.2855E 03	.3199E 00	.56276-0	1 .317.
1300.00	960.00	.2918E 03	.0244E 00	.5777E-0	1.316
1300.00	930.00	.2902E U3	.3288E 00	:5924E-0	1 .318
1399.00	1000.00	.3045E 03	.3333E 00.	.6071E-0	1320
1000.00	1820.68	.3199E 03	.3376E 00	.6214E-0	1 .321 🧋
1099.00	1040.00	.3173E 03	.3419E 00	.6357E-0	1 .023

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P (PSIA)	T (°F)	H (BTU/LB)	S (BTU/LB-°F)	V (FT ³ /IB)	CP (BTU/LB~ºE)
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• 67 21 2 1 2 12	Antar susu	a az dum türt	1 4/ 117-131	4 (2030)LT., 22.4	10 Mar 19
	00.00	.0404 <u>0</u> 01	04655-01 04655-01	100000-01	
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1000.00	120.00	.didab 96	-4389E-01 Fotor of	.10766-01	. 1. 000 1.020
1000.00	140.00	2542 <u>5</u> 06	.03166701	11306-01	1.000
1566.00	160.00	.SINE UC	, 66636-01	11746-01	1.024
1500.00	180.00	.3723E 02	./1121-01	.12108-01	. 1.120
1500.00	200.00	.4283E 82	.79861701 	.12046-01	. 1.108
1500.00	220.00	.4808E 82	.88466-01 ocost ot	.13106-01	. 1.1(C) 1.0000
1500.00	240.00	.3447E 02	.96906-01	.13715-01	1.200
1500.00	200.00	.6051E 02	.1003E 00	.14556-01	1.240
1500.00	230.00	.6671E 82	.11378 00	.10036-01	L Ás⊑íto A mam
1500.00	300.00	.,309E 82	.12296 99	.10866-01	1.1.312
1500.00	320.00	.79555 02	.13028 00	.15/05-01	L 1.000 - 4.44
1500.00	340.00	.86441 02	.1386E 00	.1.805-01	1.414
1560.00	369.00	.9351E 02	.1470E 00	.19086-01	1.478
1500.00	380.00	.10096 03	.1556E 00	.20571-0.	1.600
1569.69	400.00	.1089E 03	.16478 00	.dd/dt-01	1.780
1589.00	420.00	.1178E 03	.1745E 00	.20531-0.	1 2.200
1500.00	449.00	.1288E 03	.18521 00	.30//E-0.	L 2.040
1566.66	460.00	1445E 49	82368 48	4006E 40)######## {
1500.00	489.00	8672E 47	4978E 49	24046 40	<u>ॖ</u> ॖॖॖॖॖॖॖ॑ ग़ग़ग़ग़ग़ग़ग़ ॴॴॴॴॴॴॴॴ
1509.00	500.00	1590E 50	9126£ 49	4407E 40	********
1500.00	520.00	2313E 50	1327£ 50	6410E 40	5########
1500.00	540.00	3835E 50	1742E 50	8413E 40	3######## 3444444444
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1500.00	600.00	5203E 50	2987E 50	1446E 4	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~
1500.00	620.00	5926E 50	3401E 50	1543E 4	/#########
1500.00	640.00	6649E 50	3816£ 50	1843E 4	, ########
1500.00	660.00	/3/1E 50	4231E 00	2043E 4	
1500.00	689.99	8094E 50	4646E 00	2244E 4	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~
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1500.00	720.00	3033E 30 40025 Et	04/01 00 Fonat Fa	- 7,2544E 4 - 9044E 4	<i>;</i>
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1500.00	760.00	-,1078E 01			(######### 4 015
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APPENDIX D

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NOMOGRAM FOR SIMPLE PAYBACK PERIOD OF BOTTOMING CYCLE SYSTEM

EXPRESSIONS FOR DEVELOPMENT OF PAYBACK PERIOD NOMOGRAM

Definitions:

DC	- base diesel engine cost (\$)
50	- Dase dieser engine cost (4)
BC	= bottoming cycle cost (\$)
BCO	= bottoming cycle power output (hp)
\$/hp	= cost per additional horsepower above base diesel engine rating
BSFC.	= brake specific fuel consumption of engine (lb/hp-hr)
MPG *	= fuel economy of engine (mile/gal)

*Subscript denoting engine configuration as follows:

0	= reference diesel engine (TC/A engine from which others are keved)
D BC	<pre>= base diesel engine = compound engine (diesel + bottoming cycle)</pre>
miles/yr \$/gal	= annual vehicle mileage = fuel cost

Since the BSFC and MPG are keyed to the reference diesel engine (TC/A) where $BSFC_0 = 0.310$ lb/hp-hr and $MPG_0 = 5.75$ mile/gal, then the fuel usage of any engine configuration can be determined as follows:

• Fuel Usage of Base Diesel Engine $(gal/yr) = \frac{miles/yr}{MPG_D}$

 $= \frac{\text{miles/yr}}{\begin{pmatrix} BSFC \\ O \\ \overline{BSFC} \\ D \end{pmatrix}} (MPG_{O}).$

Likewise,

Fuel Usage of Compound Engine (gal/yr) =
$$\frac{\text{miles}/\text{yr}}{\text{MPG}_{BC}}$$

= $\frac{\text{miles}/\text{yr}}{\left(\frac{\text{BSFC}}{\text{BSFC}_{BC}}\right)}$ (MPG_o).

In both the above expressions for fuel usage,

$$(BSFC_{O})(MPG_{O}) = (0.310)(5.75) = 1.7825.$$

So:

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Fuel Savings (gal/yr) = Fuel Usage of Diesel Engine - Fuel Usage of Compound Engine



The Annual Fuel Cost Savings (\$/yr) then becomes the product of Fuel Savings (gal/yr) and Fuel Cost (\$/gal) or,

$$\frac{(\text{miles/yr})(\text{BSFC}_{D} - \text{BSFC}_{BC})(\$/\text{gal})}{1.7825}.$$

The Engine Capital Cost Difference is the sum of the Base Diesel Engine Cost and the Bottoming Cycle Cost minus the Diesel Engine Cost at the compound engine rating. This is expressed as follows:

Engine Capital Cost Difference (\$) = DC + BC - (DC + BCO x \$/hp) = BC - (BCO)(\$/hp).

<u>The Payback Period</u> is defined as the Engine Capital Cost Difference divided by the Annual Fuel Cost Savings of the compound engine and is expressed as:

Payback Period (yr) = $\frac{(BC - BCO \times \$/hp)(1.7825)}{(miles/yr)(\$/gal)(BSFC_D - BSFC_{BC})}$



Nomogram for Determination of Payback Period for a Truck Bottoming Cycle System

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1. Report No.	2. Government Accession No.	3. Recipient's Catalog No.			
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An RC-1 Organic Rankine Bot	December 1983				
Adiabatic Diesel Engine	6. Performing Organization Code				
7. Author(s)	8. Performing Organization Report No.				
L. R. DiNanno, F. A. DiBell	TE4322-251-83				
		10. Work Unit No.			
9. Performing Organization Name and Address Thermo Electron Corporation		11. Contract or Grant No.			
45 First Avenue		DEN 3-302			
P.O. Box 459	F 4	13. Type of Report and Period Covered			
Waltham, Massachusetts U22	<u> </u>				
U.S. Department of Energy		contractor Report			
Office of Vehicle and Engin	e R&D	14. Sponsoring Agency-Code Report No.			
Washington, D.C. 20585	-	DOE/NASA/0302-1			
15. Supplementary Notes	<u></u>				
Final Report. Prepared under Interagency Agreement DE-AIO1-80CS50194. Project Manager, M. Bailey, Energy Technology Division, NASA Lewis Research Center, Cleveland, Ohio 44135.					
A system analysis and prelim Rankine-cycle system to bott diesel engine. The bottomin cylindrical air-cooled conde The bottoming cycle output i compounding the reference 31 resulting brake specific fue engine. The bottoming cycle delivers a compound engine b This system for heavy-duty t RC-1, which is a mixture of percent hexafluorobenzene (H thermal stability testing of that simulates the operation operation were completed wit up to 900°F. This report de contracts awarded under the Technology Program. Related waste-heat recovery from the CR-168255 (Steam Rankine) an	inary design were cond om the high-temperatur g cycle is a compact p nser-regenerator modul s 56 horsepower at des 7 horsepower turbochar 1 consumption of 0.268 when applied to a tur rake specific fuel cor ransport applications 60 mole percent pental FB). Included in thes the RC-1 organic flut of Rankine-cycle. Mo h results showing that scribes the work perfo Department of Energy's reports in the area of exhaust of adiabatic d NASA CR-168257 (Brag	ucted for an organic e waste heat of an "adiabatic" ackage that includes a e and other unique features. ign point conditions when ged (TC) diesel engine with a bl/hp-hr for the compound bocompound (TCPD) diesel sumption of 0.258 lb/hp-hr. uses the organic working fluid luorobenzene (PFB) and 40 mole e 1983 work efforts was the d in a dynamic fluid test loop ore than 1600 hours of the RC-1 is thermally stable ormed for one of the multiple Heavy-Duty Transport of alternative power cycles for diesel engines are NASA (ton).			
17. Key Words (Suggested by Author(s)) Waste heat Fuel economy Organic Rankine bottoming c Adiabatic diesel engine	ycle DOE C	on Statement ssified-Unlimited Category 85 ategory UC-96			
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