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## The Effect of Channel Convergence on Heat Transfer in a Passage With Short Pin Fins

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# THE EFFECT OF CHANNEL CONVERGENCE ON HEAT TRANSFER 

IN A PASSAGE WITH SHORT PIN FINS

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#### Abstract

SUMMARY Results from a series of experiments are presented showing the effect of channel convergence on heat transfer in short pin fins as used in turbine blade internal cooling schemes. Array averaged heat transfer coefficients were obtained for two configurations of staggered arrays of short pin fins in a converging channel. One configuration contained pins with length to diamerer ratio of two in a constant height channel with channel sidewalls converging to give an area ratio of two to one from inlet to outlet, giving only accelarated flow. The second configuration contained pins in a constant width channel with the top and botton surfaces of the channel converging and with length to diameter ratio varying from two to one streamise through the test section. This configuration had the same area ratio from inlet to outlet as in the first configuration, giving varying piri length as well is accelerated flow. In addition to the pin fin configurations, two flat plate configurations were also tested which had the same geometries as the pin fin configurations but no pins.

Results show that for the constant height pin fin configuration, the Nusselt numhers are approximately the same as in previous tests done on straight channel test sections with pins of the same length to diameter ratio. For the constant width pin fin configuration, results indicate Nusselt numbers approximately 20 percent lower than for tests performed on configurations with constant pin length in the same range of length to diameter ratio. This suggests that the varying pin length toes have some effect on the heat transfer, but that the flow acceleration has a negligible effect. For the flat plate configurations, both indicate Nusseit numbers in the same range as is other similar flat plate tests. As in the above mentioned pin fin tests, the constant width configi ration exhibited Nusselt numbers silightly lower (about 6 percent) than for the constant height configuration. This shows that a constant widih configuration (pin fin or flat plate) exhibits a lower Nusselt number level than a constant height configuration.


## INTRODUCTION

In modern gas turbine engines, there are many different types of turbine blade cooling schemes availaile to the designer. The method of particular interest in the present study is the use of short pin fins as used in traliling edge couling passages. Until quite recently, the only information of this type avallable in the literature was for relatively lorig pins of mure than four diameters in length. Since the length of the pin firis used for gas turbine applications is generally less than four diameters, most of this information could not be used effectively.

Recently, some data specifically for applications of pin fins to gas turbines have become avallable. One program underiaken at Arizona State University (refs. 1 to 3) measured the heat transfer from short pin fins arranged in staggered 10-row (streanwise) arrays. This work (refs. 1 and 2) measured the spamwise-averaged heat transfer for both the pin and endwall surfaces on a row by row basis for pins with $L_{p} / D$ of one. These results showed that the heat transfer increased in the first four rows of the array and then decreased slightly through the rest of the array. This was similar to the trends found in long cylinder and tube bank data.

Vanfossen (ref. 4) measured the heat transfer from both pin and endwall surfaces for several staggered four-row (streamise) arrays of si.ort pin fins with $L_{p} / D$ of one half and two. It was found that for this range of length to dianeter ratio, all the data could be represented by a single curve. It was also found that the heat transfer coefficients on the pin fin surface were about 35 percent higher than on the endwall surface.

Another set $0^{\text {: }}$ experiments determined the effect of position within an array on the reat transfer tc a single pin with $L_{p} / D$ of three (ref. 5). The results here indicated that the heat transfer to a single pin in a staggered array increased significantly for one, two or three rows of pins added upstream of the test pin. For four or more rows added upstream, the heat transfer decreased slightly. This trend is similar to that found in references 1 and 2. It was also determined that the average channel velocity should be used as the reference velocity in the Reynolds number.

In reference 6 the effect of the number of rows and the length to diameter ratio on the heat transfer was investigated. It was determined that the number of streamise rows has only a slight effect on the heat transfer to short pin fins. The dominant effect, however, was the length to diameter ratio, with $L_{p} / D$ of four giving much higher heat transfer than was found in references 1, 2 and 4 for shorter pins. Reference 6 also showed that by using the average channel velocity and the proper definition of the characteristic length, the data of references 1 and 2 could be made to fall on the correlation derived in reference 4.

Reference 7 presents heat transfer results for several short pin fin configurations in which the length to diameter ratio varied streamwise through the length of the test section. However, this work does not specifically address the current objective and left unresolved which effect was dominant, channel convergence with the resulting accelerating flow or varying length to diameter ratio.

This paper presents the resilts of a series of tests designed to separate the effects of varying pin length from channel convergence. Four test sections were fabricated to give an area ratio from inlet to outlet of two to one. One test section contained pins in a constant height channel with length to diameter ratio of two with the channel sidewalls converging. Another test section contained pins in a constant width channel with the top and boitom channel surfaces converging and with length to diameter ratio varying from two to one streamwise through the test section. The pins were all spaced 4 diameters spart in an equilicteral triangular array with four rows in the streamwise direction, as in previous work done at NASA (refs. 4 and 6). Two other test sections were fabricated which were identical to the above two test sections
but contained only flat plate surfaces (no pin fins). Heat transfer results are presented in the form of Nusselt number versus Reynolds number over a range of Reynolds numbers of interest for this applicatior. Results are compared to those of references 4 and 6 .

## NOMEMCLATURE

$A^{\prime} \quad$ average flow area, $V / L, \mathrm{~cm}^{2}\left(1 \mathrm{n} .^{2}\right.$ )
$A_{h} \quad$ surface area of heaters, $2 \mathrm{LH}, \mathrm{cm}^{2}\left(\mathrm{in} .^{2}\right.$ )
$A_{p} \quad$ pin cross sectional area, $\pi 0_{0}^{2} / 4, \mathrm{~cm}^{2}\left(i n .^{2}\right)$
$A_{w} \quad$ endwall area per pin, $(\sqrt{3 / 2})\left(x_{t} D_{0}\right)^{2}, \mathrm{~cm}^{2}\left(1 n^{2}\right)$
$D^{\prime} \quad$ characteristic length, $\mathrm{cm}(i n$.
$D_{0} \quad$ pin diameter, cm (in.)
$h^{\prime}$ average heat transfer coefficient in pin fin test section, $\mathrm{W} / \mathrm{m}^{2} \cdot K$ (Btu/hr•ft ${ }^{2} \cdot{ }^{\circ} \mathrm{R}$ )
$h_{\text {eff }} \quad$ effective heai transfer coefficient, $W / m^{2} \cdot K$ (Btu/hr•ft ${ }^{2} \cdot{ }^{\circ} R$ )
$h_{p} \quad$ average heat transfer coefficient on pin surface, $H / m^{2} \cdot K$
( $8 \mathrm{tu} / \mathrm{hr} \cdot \mathrm{ft} \mathrm{t}^{\circ}{ }^{\circ} \mathrm{R}$ )
$h_{w}^{\prime} \quad$ average heat transfer coefficient on endwall surface, $W / m^{2} \cdot K$ (Btu/hr•ft ${ }^{2}{ }^{\circ} R$ )
$k_{a} \quad$ thermal conductivity of air, w/m•K (Btu/hr•ft•• $R$ )
$k_{p} \quad$ thermal conductivity of pin material, W/m•K (Btu/hr•ft•*R)
L test section streamilise lengtr, cm (in.)
$L_{p} / D \quad$ pin length to diameter ratio
${ }_{1}^{p}$ pin half length, cm (in.)
$m \quad$ fin parameter, $\sqrt{h^{\prime} P / k_{\rho} A_{p}}$
$\mathrm{Nu} \quad$ Nusselt number, $h^{\prime} D^{\prime} / k_{a}$
$\mathrm{P} \quad$ pin perimeter, $\pi \mathrm{D}_{\mathrm{o}}$, cm (in.)
$Q$ heat dissipated in electric heater, $W$ (Btu/hr)
Re Reynolds number, (w/A')D'/
$S \quad$ total heat transfer surface area, $\mathrm{cm}^{2}\left(\mathrm{in}^{2}{ }^{2}\right)$
$T_{a w}$ adiabatic wall temperature, $K\left({ }^{\circ} R\right)$
${ }_{1} \quad$ Eckert reference temperature, $K\left({ }^{\circ} R\right)$
${ }^{\prime}$ s static temperature, $K\left({ }^{\circ} R\right)$
$i_{w} \quad$ endwall temperature, $K\left({ }^{\circ} R\right)$
$V^{W}$ flow channel vclume minus pin volume, $\mathrm{cm}^{3}$ (in. ${ }^{3}$ )
W test section width, cm (in.)
$w \quad$ mass flow rate, $\mathrm{kg} / \mathrm{s}(1 \mathrm{bm} / \mathrm{s})$
$x_{s} \quad$ ratio of pin haight to pin diameter, (same as $L_{p} / D$ )
$x_{t} \quad$ ratio of pin spacing to pin diameter, transverse direction
$\mu \quad$ viscosity of air. kg/m.s (lbm/ft.s)

## DESCRIPTION OF EXPERIMENT

## Test Section and Flow Apparatus

A schematic of the rig used for the pin fin experiments is shown in figure 1. Room air was drawn through a constant acceleration inlet, into the test section and then through either an orifice flow meter (large flow rates) or through a venturi flow meter (small flow rates). The air passed through the respective flow control valves and then on into the laboratory altitude exhaust system. Air temperatures were measured at the test section inlet, test section outlet, and ilow meters. Air density at the flow meters was calculated using the ideal gas law. The data were collected and converted to engineering units using the laboratory data collection system (ref. 8).

Figure 2 shows the test section assembly used for all the pin fin experiments. Shown in the figure are the inlet, endwall, and outlet thermocouples. The top and bottom surfaces of the inlet were covered with fine grit sandpaper and a 0.519 cm ( 0.063 in. ) diameter trip wire was attached to the surface in order to ensure turbulent flow. The test section was enclosed in clothreinforced pheno.ic plastic and the entire assembly was wrapped in fiberglass insulation in order to minimize heat losses.

Figures 3 and 4 show schematics of the two pin fin test sections used, illustrating the channel geometry. Figures 5 and 6 show the assembled pin fin test sections, viewed in the direction of flow. These pictures show the difference in the convergence between the constant height configuration and the constant width configuration. The two flat plate test sections were identical in geometry but have no pin firs. Each test section was made of two 0.635 cm ( 0.250 in.) thick copper plates, with holes drilled for the insertion of the pins for the two pin fin test sertions. The pins, which extended completely through the endwall plates, were soldered into place. The dimensions of the flow channels for all configurations are given in table $I$.

The test sections were heated with commercially avallable electric foil resistance heaters attached externally to the endwalls with pressure sensitive adhesive. The streamwise centerline temperatures of the endwalls were measured by thermocouples inserted into holes drilled along the edge of each endwall plate. The thermocouple connections and the electric heaters can be seen in figures 5 and 6.

## Test Procedure

In order to account for heat losses, a set of calibrations was made for each test section with no air flow through the test section. By measuring the endwall temperatures, the ambient air temperature, and the total heat flux, an overall heat loss coefficient was determined for each configuration. This was done after the test section had reached a steady state condition, which was

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usually overnight. During a data run, the Reynolds number was changed by varying the flow rate through the test section. The temperature of the endwall plates was kept constant using an electronic heater controlling system which regulated the power to each set of heaters.

## Oata Analysis

The theoretical modei used as the basis for the pin fin data analysis is shown in figure 7. The total heat flux for the pin and endwall surfaces is the sum of the heat flux through the endwall and the heat flux through the pin. The average heat transfer coefficients on the pin surface and endwall surface are $h_{p}^{\prime}$ and $h_{w}^{\prime}$ respectively. The heat lost by the pin fin surface (endwalls included) is equated to the heat lost by a plain surface with heat transfer coefficient heff. For the purpose of this work, the average heat transfer coefficient on the pin, $h_{p}^{\prime}$, is assumed equal to the heat transfer coefficient on the endwalls, $h_{w}$, and is denoted by $h^{\prime}$.

The overall loss heat transfer coefficient, hloss. is defined as

$$
\begin{equation*}
h_{\text {loss }}=\frac{Q_{\text {loss }}}{A_{h}\left(T_{w}-T_{d}\right)} \tag{1}
\end{equation*}
$$

This loss coefficient was found over a range of temperature differences, $T_{w}-T_{a}$, and a least squares curve fit was applied to the data with temperature as the independent variable. Thus, for a given temperature difference a corresponding loss coefficient could be found and the heat loss from the test section could be calculated.

The effective heat transfer coefficient, $h_{e f f}$, is defined as

$$
\begin{equation*}
h_{e f f}=\frac{\left(0-0_{\operatorname{loss}}\right)}{A_{w}\left(T_{w}-T_{a w}\right)} \tag{2}
\end{equation*}
$$

where $T_{a w}$ is the adiabatic wall temperature calculated from

$$
\begin{equation*}
T_{a w}=T_{s}+r\left(T_{t}-T_{s}\right) \tag{3}
\end{equation*}
$$

with $r$, the recovery factor, taken as the square root of the Prardtl number. The total temperatures were found by ma!ing an energy balance through the channel. The static temperatures were found using the compressible gas relationship between total and static temperatures.

Equation 2 gives the effective heat transfer coefficient, including the effects of pin efficiency. This can be rewritten as

$$
\begin{equation*}
Q_{\text {total }}=n_{e f f} A_{w}\left(T_{w}-T_{a w}\right) \tag{4}
\end{equation*}
$$

The heat flux through the endwalis is given by

$$
\begin{equation*}
Q_{w a 11}=h^{\prime}\left(A_{w}-A_{p}\right)\left(T_{w}-T_{a w}\right) \tag{5}
\end{equation*}
$$

where $A_{W}$ is the endwall area and $A_{p}$ is the cross-sectional area of the pin. The heat flux through the pin only is given by

$$
\begin{equation*}
Q_{p i n}=\sqrt{P h^{\prime} k_{p} A_{p}} \tanh (m 1)\left(T_{w}-T_{a w}\right) \tag{6}
\end{equation*}
$$

with $p$ being the perimeter of a pin, $l$ being the half length of the pin, and the fin parameter $m$ being defined by

$$
\begin{equation*}
m=\sqrt{\frac{h^{\prime} p}{k_{p} A_{p}}} \tag{7}
\end{equation*}
$$

The pin conductivity, $k_{p}$, was taken ps $3.46 \mathrm{~W} / \mathrm{cm} \cdot \mathrm{K}$ ( $200 \mathrm{Btu} / \mathrm{hr} \cdot \mathrm{ft} \cdot{ }^{\circ} \mathrm{R}$ ). Then the energy balance yiolds the following equation:

$$
\begin{equation*}
h^{\prime}\left(1-\frac{A_{p}}{A_{w}}\right)+\left(\frac{\sqrt{P h^{\prime} k_{p} A_{c}}}{A_{w}}\right) \tanh (m 1)-n_{e f f}=0 \tag{8}
\end{equation*}
$$

The above transcendental equation was solved iteratively for $h$ ' for each measured value of $h_{\text {eff }}$.

For the flat plate test sertions, an overall ioss heat transfer coefficient was found as described coove. Then since there were no pins, there was no need for an iterative procedure to solve for $h '$. The heat transfer coefficient is calculated as

$$
\begin{equation*}
n^{\prime}=\frac{\left(0-Q_{\operatorname{los} s}\right)}{A_{w}\left(T_{w}-T_{a w}\right)} \tag{9}
\end{equation*}
$$

with $h^{\prime}$ being the average heat transfer coefficient on the endwall and sidewall surfaces in the plain channel.

The data was put in dimensionless form of Reynolds number and Nusseit number for purposes of comparison. The Reynolds number is defined as

$$
\begin{equation*}
\operatorname{Re}=\left(\frac{W^{\prime}}{A^{\prime}}\right) \frac{D^{\prime}}{\mu} \tag{10}
\end{equation*}
$$

where

$$
\begin{align*}
& D^{\prime}=\frac{4 V}{S}  \tag{11}\\
& A^{\prime}=\frac{V}{L} \tag{12}
\end{align*}
$$

and $\mu$ is the viscosity of air. The Nusselt number is defined as

$$
\begin{equation*}
N u=\frac{h^{\prime} 0^{\prime}}{k_{a}} \tag{13}
\end{equation*}
$$

where $k_{a}$ is the thermal conductivity of air. Both $\mu$ and $k_{a}$ were found from curve fits of data from reference 9 and were evaluated at the Eckert reference temperature (ref. 10), defined as

$$
\begin{equation*}
T_{r}=0.5 T_{W}+0.28 T_{S}+0.22 \mathrm{~T}_{\mathrm{aw}} \tag{14}
\end{equation*}
$$

The experimental uncertainty based on ine method of Kline and McClintock (ref. 11) is less than three percerit for the Reunolds number and less than four percent for the Nusselt number.

## RESULTS AND DISCUSSION

Array averaged heat transfer coefficients for both pin and endwall surfaces were found for a range of flow rates and are presented in the form of Nusselt number versus Reynolds number for two pin fin configurations and two flat plate configurations in a converging channel. Figure 8 is a comparison of the constant height pin rin data wín the results of references 4 and 6 . The constant height data with $L_{p} / D$ of two exhibits approximately the same Nusselt numbers as the correlation derived in reference 4 for $L_{p} / D$ of less than or equal to two. In addition, the constant height data with $L_{p} / D$ of two falls approximately 20 percent below both the four and eight row data of reference 6 with $L_{p} / D$ of four. This difference is to be expected based on the results of reference 6 . The effect of the channel convergence on the heat transfer to short pin fins of constant length to dianoter ratio is merely a slight change in the slope of the data and virtually no change in the Nusselt number levels.

Figure 9 shows both the constant height data and the constant width data plotted with the correlation from reference 4 . This correlation is for all short pin fins with length to diameter ratio between one half and two. It can be seen from figure 9 that the constant width data with $L_{p} / 0$ of two to $L_{p} / D$ of one (the same range as in reference 4) fall approximately 20 percent below the correlation from reference 4. This is significant since the length to diameter ratios for these two sets of data are in the same range. One would have expected these two sets of data to be identical since the $L_{p} / D$ are in the same range. Thus the changing pin length must be a significant factor. A least squares curve fit of the constant width data is

$$
\begin{equation*}
N u=0.147 \operatorname{Re}^{0.667} \tag{15}
\end{equation*}
$$

Where the Nusselt number and Reynolds number are based on the characteristic length $D^{\prime}$. Since the area ratios in both converging pin fin configurations are the same (two to one from inlet to outlet), it is apparent that in the constant width configuration the changing pin length has a significant effect on the Nusselt number level.

Figure 10 shows the current pin fin data and the current flat plate data plotted with the results of reference 4 . It can be seen that both the constant height and the constant width flat plate data fall at approx!mately the same level as the nonconverging flat plate data from reference 4. This tends to indicate that these results are not rig dependent since the data from reference 4 were obtained in the same experimental factlity. This also shows that there is not much effect on Nusselt number levels in converging configurations without pin fins. Also, the basic trend of the two sets of data are the same with the constant height configurations having consistently higher Nusselt number levels than the constant width configurations. This would tend to indicate that there may be some effect due to the direction of the convergence,
but this was not investigated in this program. Any further attempt to explain the reasons for the specific heat transfer characteristics in greater detall would necessitate a much more in-depth study of the flow field as well as measurements of the local pin and endwall heat transfer.

## SUMmARY OF CONCLUSIGNS

The results of this investigation of the effect of channel convergence on the heat transfer in short pin fins are as follows:

1. The constant height data with $L_{p} / D$ of two exhibited Nusselt number levels approximately the same as the correlation derived by Vanfossen, with a silightly different slope, however. Since the $L_{p} / D$ is in the same range as the correlation of Vanfossen, the effect of flow acceleration due to channel convergence in cases of constant $L_{p} / D$ is negligible.
2. The constant width data with $L_{p} / D$ of two to $L_{p} / D$ of one falls approximately 20 percent lower than the correlation derived by VanFossen for the same range of $L_{p} / 0$. It appears that the varying pin length has a definite effect on the heat transfer characteristics.
3. For tests on similar converging configurations using flat plates only (no pin fins), the flat plate data seems to have the same general behavior as nonconverging flat plates. This indicates no rig dependent behavior in the data.
4. The constant height configurations (buth pin fin and flat plate) fall consistently higher than the constant width configurations. Therefore, there may be some effect due to the direction of the convergence. However, this effect was not investigated in the present program.

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TABLE I. - TEST SECTION DIMENSIONS

| Configuration | $x_{t}$ | $x_{s}$ | Length, <br> cm | Inlet |  | Exit |  |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | Width, <br> cm | Height, <br> cm | Width, <br> cm | Height, <br> cm |  |
| Constant <br> height | 4 | 2 | 9.58 | 12.7 | 1.27 | 6.35 | 1.27 |
| Constant <br> width | 4 | $2-1$ | 9.58 | 12.7 | 1.27 | 12.7 | 0.64 |

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Figure 1. - Pin lin heat transter rig.


Figure 2. - Pin fin test section assembly.

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Figure 3. - Schematic of constant height test section.


Figure 4. Schematic of constant width test section.

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Figure 5. - Constam height test section,


Figure 6 . Constant widt test section.

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$Q_{\text {TGTAL }}=Q_{\text {WALL }}+Q_{\text {FIN }}$
$Q_{\text {TOTAL }}=h_{w}^{\prime}\left(A_{w}-A_{p}\right)\left(T_{w}-T_{a w}\right)+\sqrt{P_{p}^{\prime} k_{p} A_{p}}$ ianh (mi) $\left(T_{w}-T_{a w}\right)$
$Q_{\text {TOTAL }}=h_{\text {eff }} A_{w}\left(T_{w}-T_{a w}\right)$
Figure 7. - Theoretical m.jdel.


Figure 8. - Sonstant height data plotted with results of refs. 4 and 6.

$$
\begin{aligned}
& \text { ORIGRIN } \\
& \text { OF POOR Qua }
\end{aligned}
$$


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Figure 10 - Current pin fin data plotie ' with current flat plate data.

