

NASA Technical Memorandum 86855

Thermal Conductance of Pressed Aluminum and Stainless Steel Contacts at Liquid Helium Temperatures

L.J. Salerno, P. Kittel, F.E. Scherkenbach
and A.L. Spivak

(NASA-TM-86855) THERMAL CONDUCTANCE OF
PRESSED ALUMINUM AND STAINLESS STEEL
CONTACTS AT LIQUID HELIUM TEMPERATURES

(NASA) 17 p

CSSL 20D

N87-19648

Unclas

G3/34

43547

February 1986

NASA

National Aeronautics and
Space Administration

Thermal Conductance of Pressed Aluminum and Stainless Steel Contacts at Liquid Helium Temperatures

L. J. Salerno,
P. Kittel,
F. E. Scherkenbach, Ames Research Center, Moffett Field, California
A. L. Spivak, Trans-Bay Electronics, Inc., Richmond, California

February 1986



National Aeronautics and
Space Administration

Ames Research Center
Moffett Field, California 94035

THERMAL CONDUCTANCE OF PRESSED ALUMINUM AND STAINLESS STEEL
CONTACTS AT LIQUID HELIUM TEMPERATURES

L. J. Salerno*, P. Kittel*, F. E. Scherkenbach*

NASA-Ames Research Center
Moffett Field, CA

and

A. L. Spivak†

Trans-Bay Electronics, Inc.
Richmond, CA

ABSTRACT

The thermal conductance of aluminum and stainless steel 304 sample pairs with surface finishes ranging from 0.1 to 1.6 μm rms roughness has been investigated over a temperature range from 1.6 to 6.0 K. The thermal conductance follows a simple power law function of temperature, with the exponent ranging from 0.5 to 2.25, increases asymptotically with increasing applied force, and exhibits an anomaly for surface finishes in the 0.4- μm region.

INTRODUCTION

Accurate thermal models are crucial for optimum cryogenic instrument design, particularly concerning infrared instruments and focal planes whose performance is temperature-dependent. Instruments aboard space projects such as the infrared Astronomical Satellite (IRAS), The Space Infrared Telescope Facility (SIRTF), and the Large Deployable Reflector (LDR) fall into this category. Present models are limited by a lack of cryogenic temperature data to allow one to predict the thermal conductance of the bolted joints typically used in the instrument-to-system interface. The performance of OFHC copper and brass contact pairs has already been characterized,¹⁻³ and the theory, apparatus, and experimental method have been examined in detail.³ The present work examines the thermal contact

*Research Scientist
†Research Technician

conductance of pressed pure aluminum and stainless steel 304 contact pairs having surface finishes of 0.1, 0.2, 0.4, 0.8, and 1.6 $\mu\text{m rms}$ at temperatures from 1.6 to 6.0 K, under applied contact forces up to 670 N.

RESULTS

The experimental data were fitted to a simple power law where the thermal conductance is given by

$$k = \alpha T^n$$

where k is the thermal conductance, and α and n are constants which are determined empirically.

Figures 1-9 present results obtained for aluminum sample pairs. In Figures 1-5, the thermal conductance is plotted versus temperature for each of the tested surface finishes, with applied force as a parameter. It should be noted that thermal conductance is given in units of mW/K , in keeping with the finding of Berman⁴ that thermal conductance is independent of the contact area, and dependent on the applied force. The reason for this is that the actual contact area between two surfaces is dependent on the applied force, rather than the apparent surface area. Some work-hardening of the surface occurs, which determines the contact area.

Figure 6 shows the relation between thermal conductance and applied force for the different finishes. The dependence of thermal conductance, and n on surface finish for the range of applied forces is shown in Figures 7, 8, and 9, respectively.

Figures 10-18 represent the same relations as Figures 1-9 but pertain to the stainless steel samples. Figures 19 and 20 for brass samples tested previously³ correspond to Figures 8 and 9 for aluminum and Figures 17 and 18 for stainless steel.

DISCUSSION

When Figures 1-5 and 10-14 are examined it can be seen that, as verified earlier with copper^{1,2} and brass³ samples, the thermal conductance of the aluminum and stainless steel sample pairs increases according to a power law relation with increasing applied force. The variation of conductance with applied force appears to be asymptotic, as shown in Figures 6 and 15 for data at 4.2 K, and consistent with effects observed by Berman.⁴ Figures 7 and 16 show the anomalous thermal conductance peak of the 0.4- μm surface finish sample pair at an applied force of 670 N, a behavior which was observed earlier.¹⁻³ Of particular interest is the fact that for the aluminum, the thermal conductance, instead of peaking at 0.4 μm , is actually lowest at that finish, an effect opposite to that observed in the other materials.

Figures 8, 9, 17, 18, and 19 show that for the most part, the anomaly is also apparent in both the values of α and n , the exception being Figure 20, suggesting that thermal energy transport is altered at the 0.4- μm boundary. Since the samples were lapped to achieve the required surface finish, it was thought that the 0.4- μm sample pairs could have been prepared

differently. All contact surfaces were examined under a 10X microscope to determine whether any particular characteristic of the preparation (e.g., lap mark orientation) was different for the samples in question. Specifically, it was hypothesized that if the two samples were lapped such that when they were in contact the direction of lapping coincided, the thermal conductance would be higher for this particular pair than for others. Alignment of the pairs, with respect to microscratches was examined as well. All contact surfaces showed that visible surface effects had no observable correlation with measured thermal conductance.

Since thermal energy transport is by phonons, the effect could possibly be related to the vibrational energy. It was thought that the frequency of vibration could be related to the surface roughness. If the wavelength corresponded to the wavelength of the surface asperities (i.e., the finish), perhaps this phenomenon would account for the increase in thermal conductance, an effect similar to the relative ease of driving a mechanical system at its natural frequency. An attempt was made to correlate the vibrational energy to the thermal energy, but again no causal relationship was established. In this case, the wavelength of the surface asperities would shift downward with increasing temperature, an effect which was not observed within the range of experimental error.

Since the problem of thermal contact conductance is basically one of a mismatch of the acoustic impedance between two solids, Kapitza conductance may offer some explanation of the effects observed.⁵

CONCLUSION

As found earlier with copper and brass samples, the thermal conductance of pressed aluminum and stainless steel sample pairs increases according to a simple power law function of temperature. Thermal conductance also increases asymptotically with increasing applied contact force and is related to the surface finish of the sample. The maximum contact conductance for materials other than aluminum is obtained for samples having an rms surface roughness of 0.4 μm . For aluminum, the conductance is lowest for this surface finish. Reasons for this anomalous behavior are not understood at this time. Further work is needed to identify the exact mechanism of conductance at the contact interfaces.

REFERENCES

1. Salerno, L. J., Kittel, P., and Spivak, A. L., "Thermal Conductance of Pressed Copper Contacts at Liquid Helium Temperatures," AIAA J., Vol. 22, No. 12 (1984), 1810.
2. Salerno, L. J., Kittel, P., and Spivak, A. L., "Thermal Conductance of Pressed OFHC Copper Contacts at Liquid Helium Temperatures," Thermal Conductivity 18, Proceedings of the 18th International Thermal Conductivity Conference, 187 (1985), Plenum Press.
3. Salerno, L. J., Kittel, P., Brooks, W. F., Spivak, A. L., and Marks, W. G., "Thermal Conductance of Pressed Brass Contacts at Liquid Helium Temperatures," to be published in Cryogenics.

4. Berman, R., "Some Experiments on Thermal Contact at Low Temperatures," *J. Applied Phys.*, Vol. 27, No. 4 (1956), 318.
5. Snyder, H. S., "Heat Transport through Helium II: Kapitza Conductance," *Cryogenics*, April (1970), 89.

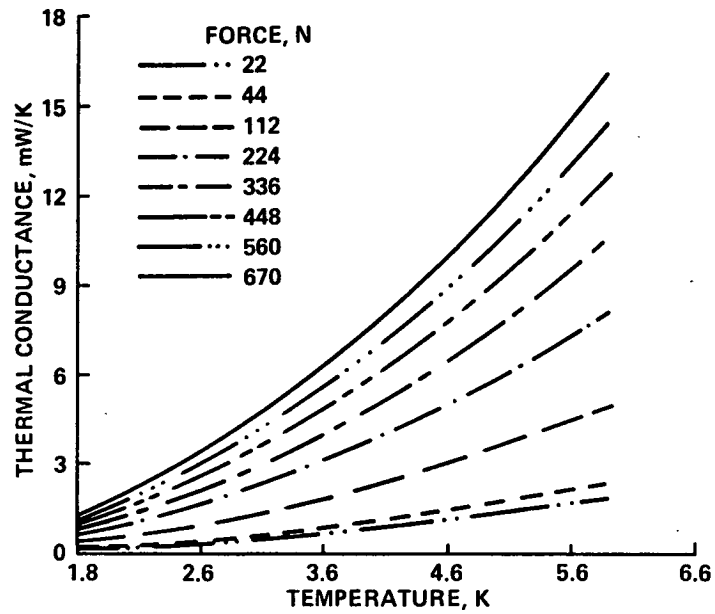


Fig. 1. Aluminum, 0.1- μm surface finish.

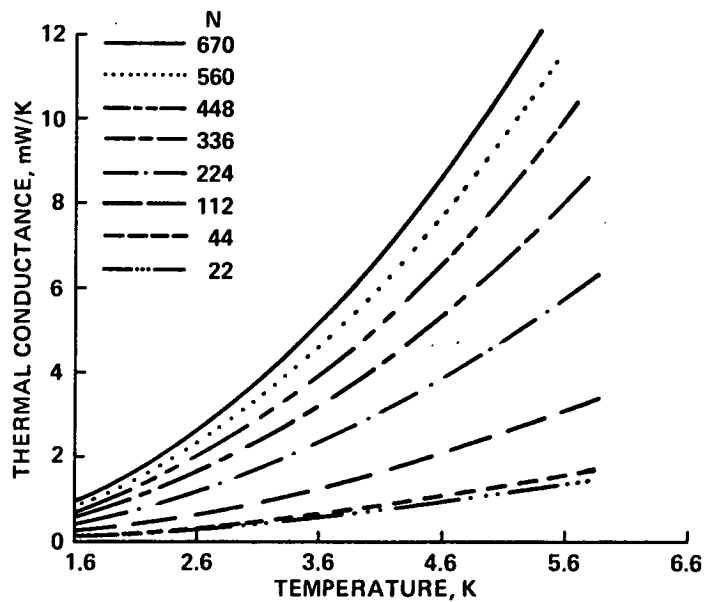


Fig. 2. Aluminum, 0.2- μm surface finish.

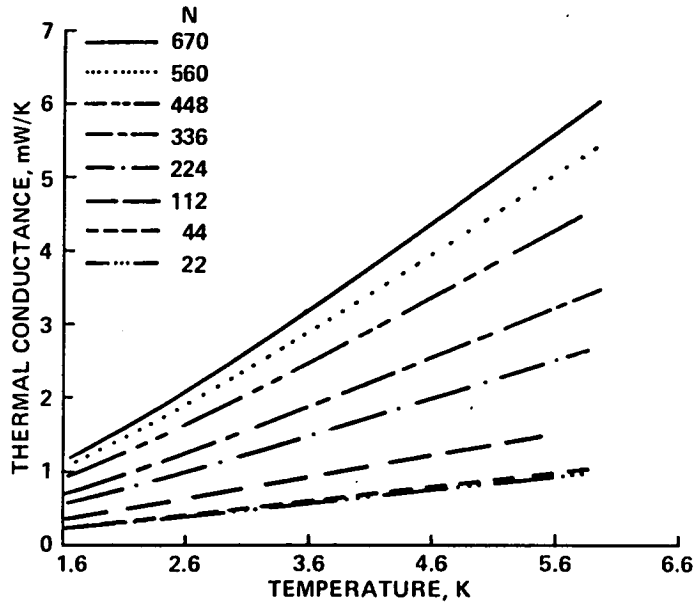


Fig. 3. Aluminum, 0.4- μm surface finish.

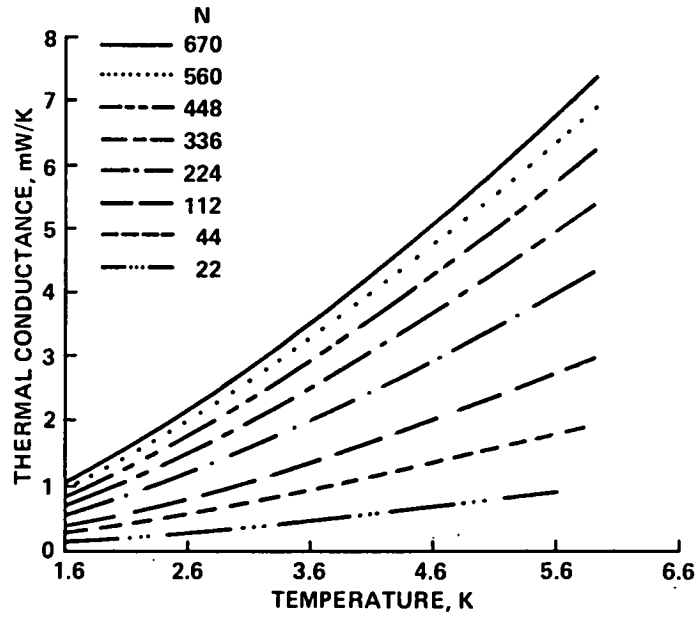


Fig. 4. Aluminum, 0.8- μm surface finish.

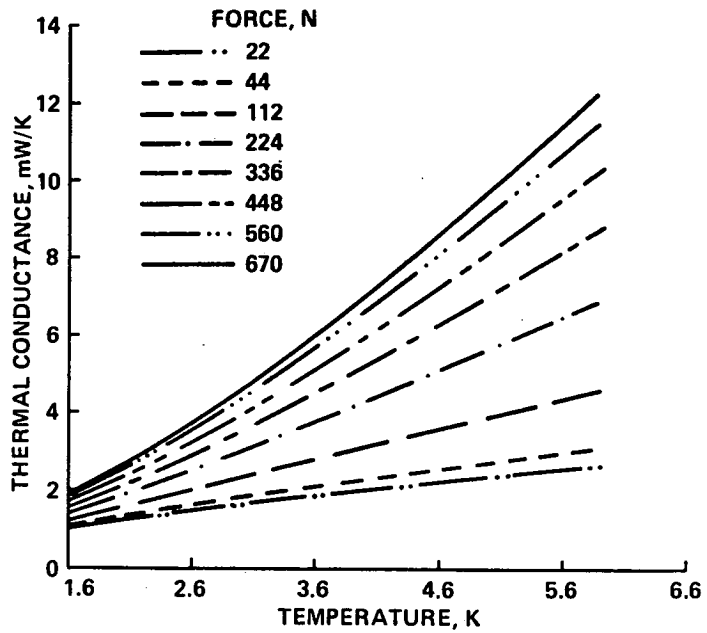


Fig. 5. Aluminum, 1.6- μm surface finish.

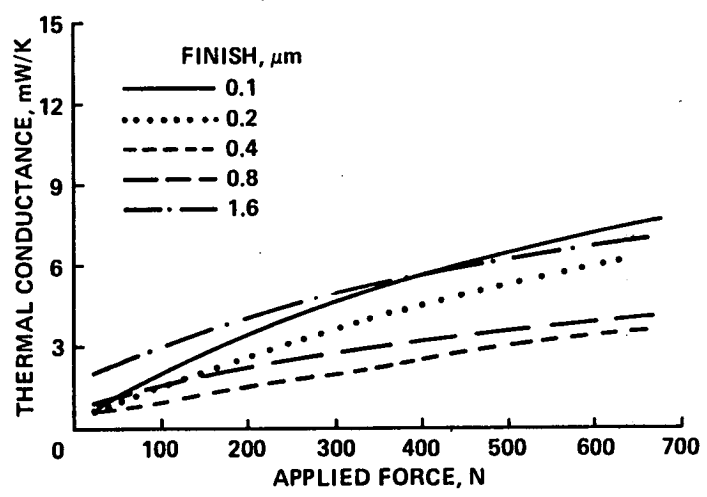


Fig. 6. Aluminum, $T = 4.2$ K.

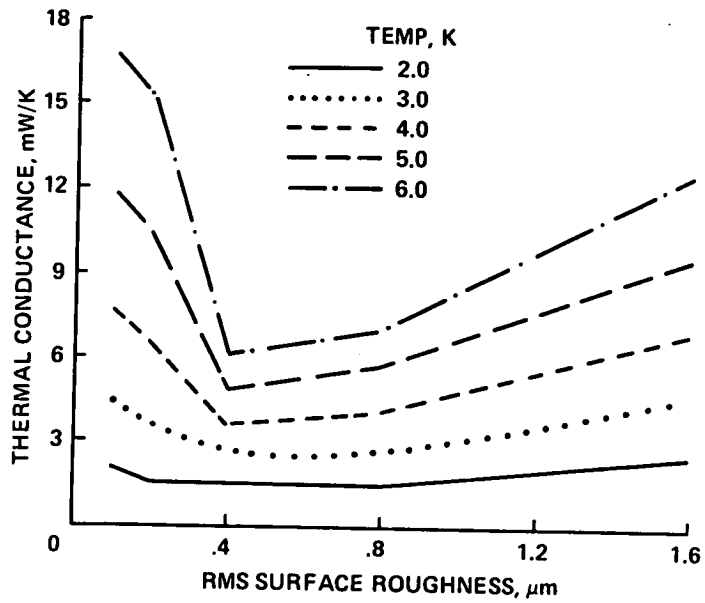


Fig. 7. Aluminum, $F = 670$ N.

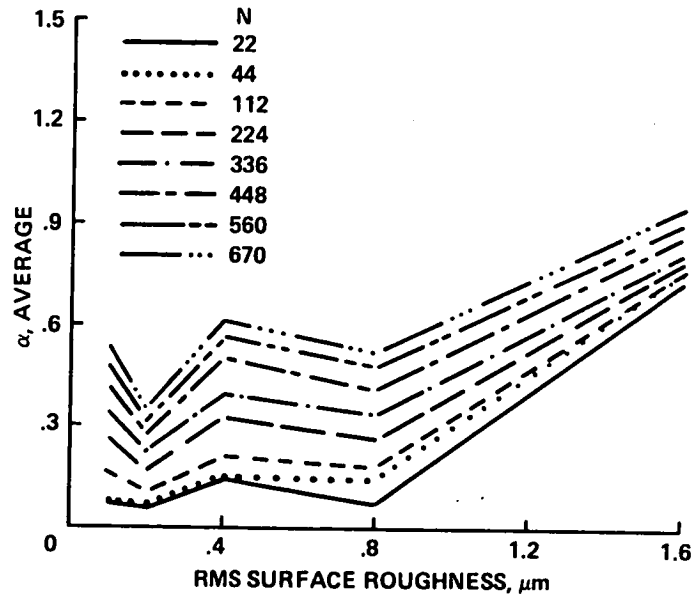


Fig. 8. Aluminum, $T = 4.2$ K.

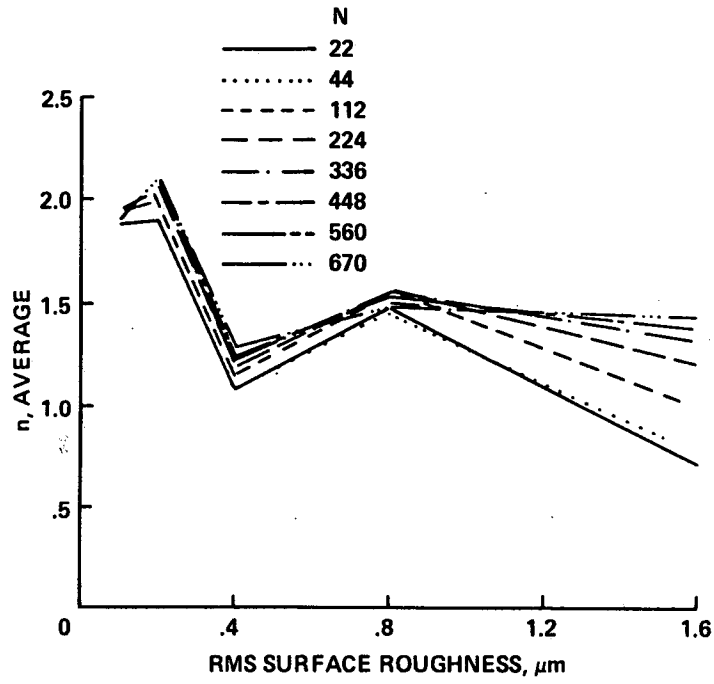


Fig. 9. Aluminum, T = 4.2 k.

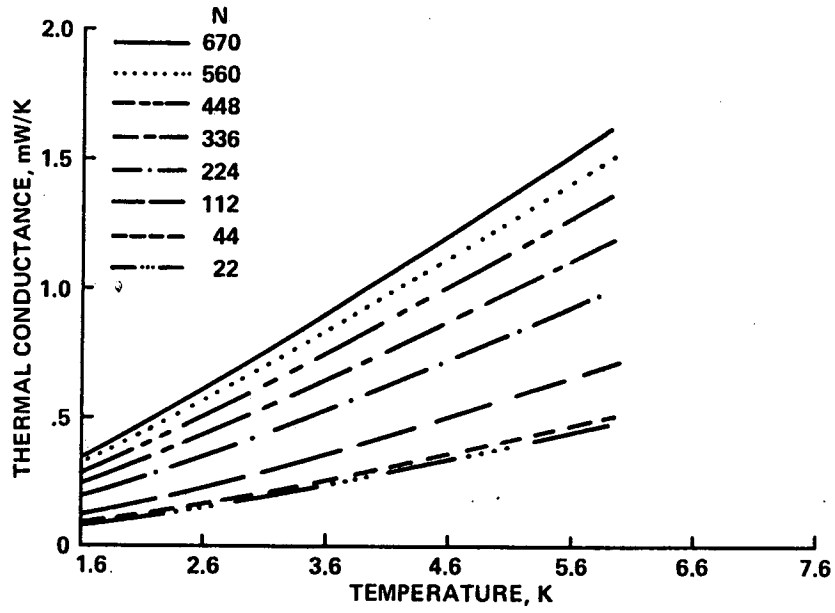


Fig. 10. Stainless steel, 0.1- μ m surface finish.

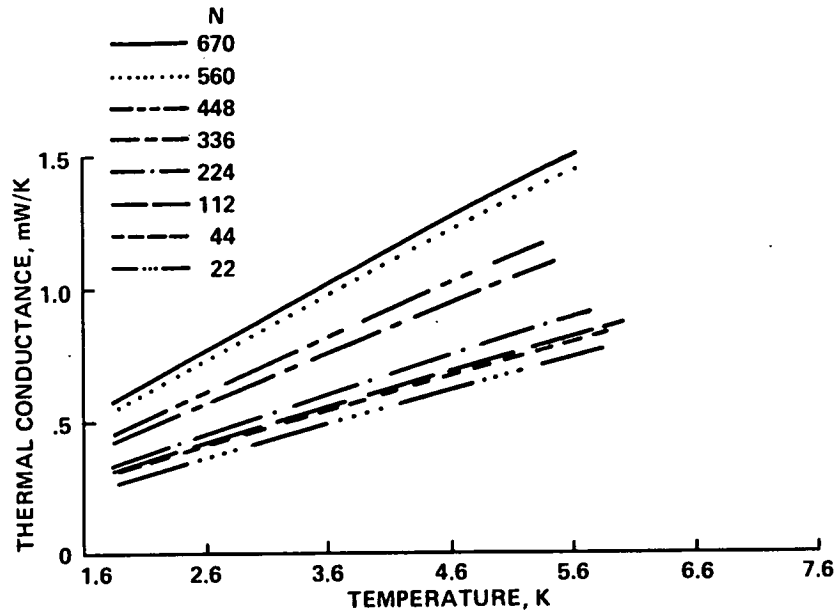


Fig. 11. Stainless steel, 0.2- μm surface finish.

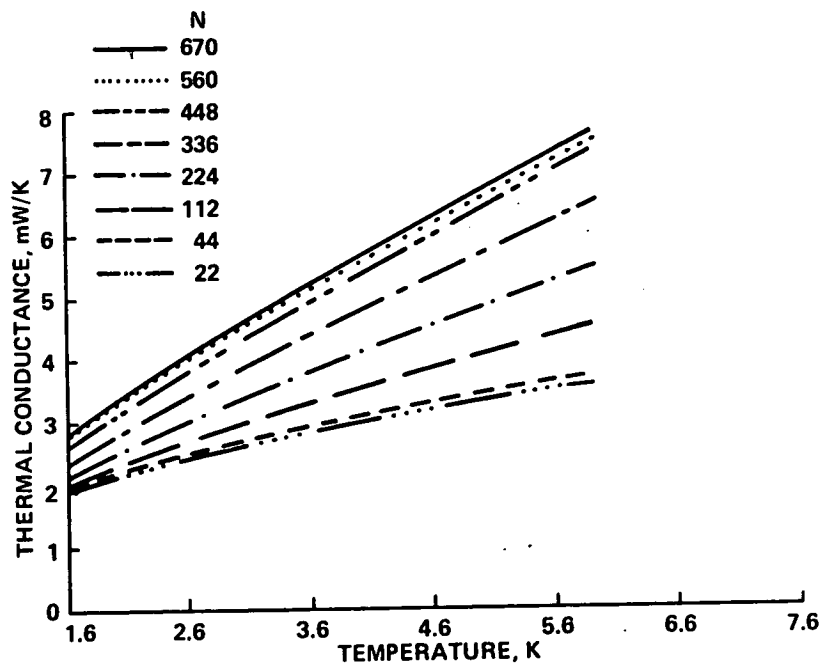


Fig. 12. Stainless steel, 0.4- μm surface finish.

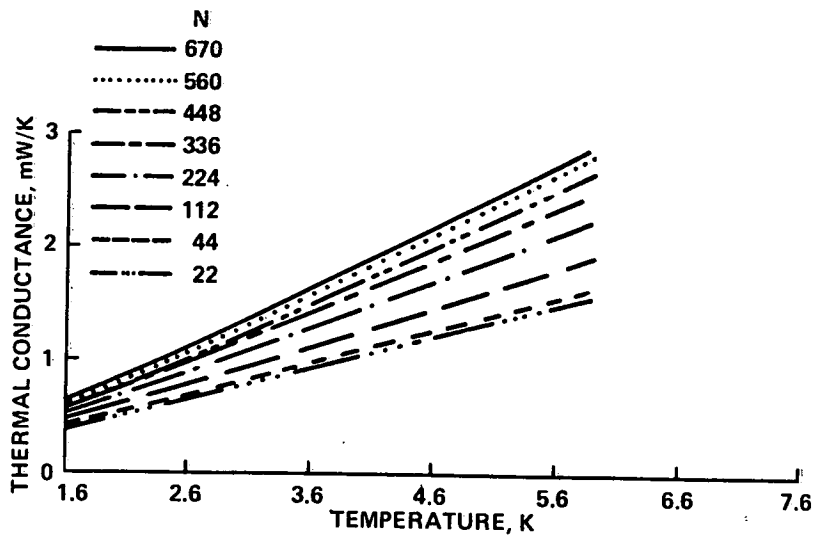


Fig. 13. Stainless steel, 0.8- μm surface finish.

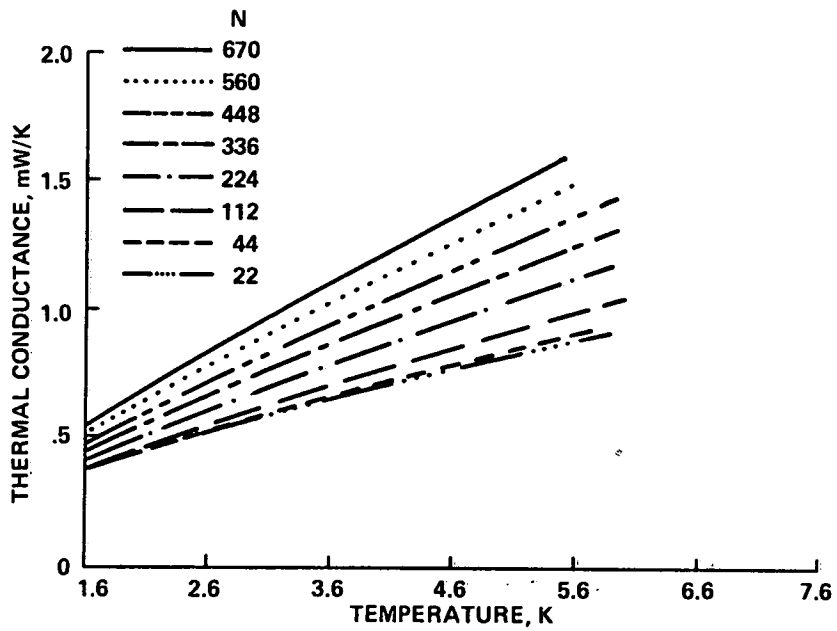


Fig. 14. Stainless steel, 1.6- μm surface finish.

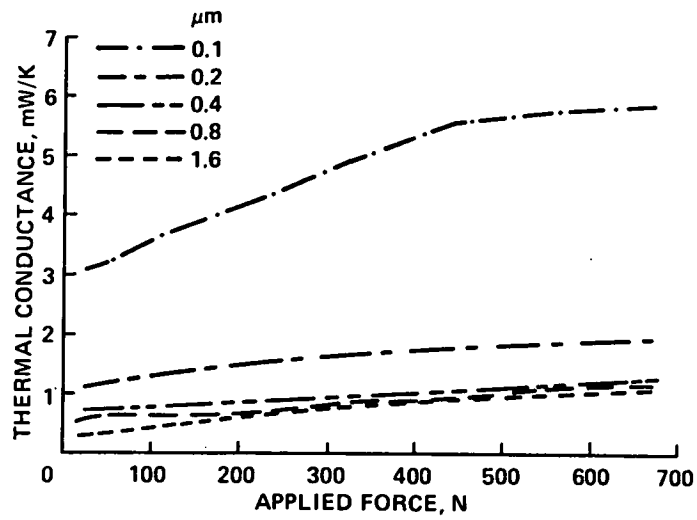


Fig. 15. Stainless steel, $T = 4.2$ K.

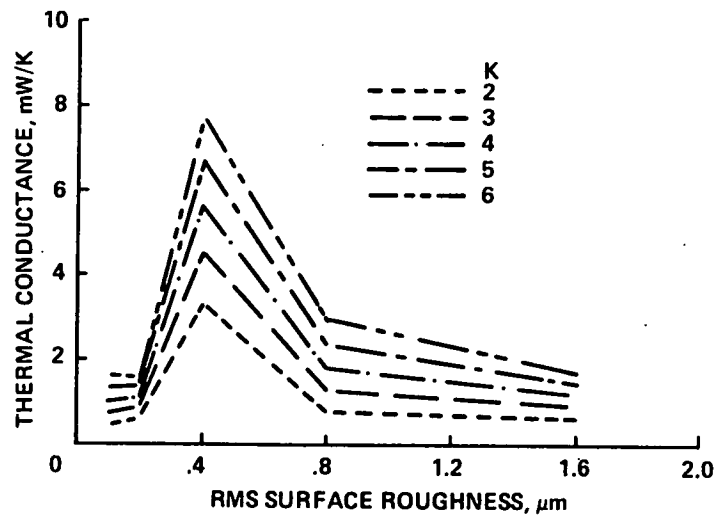


Fig. 16. Stainless steel, $F = 670$ N.

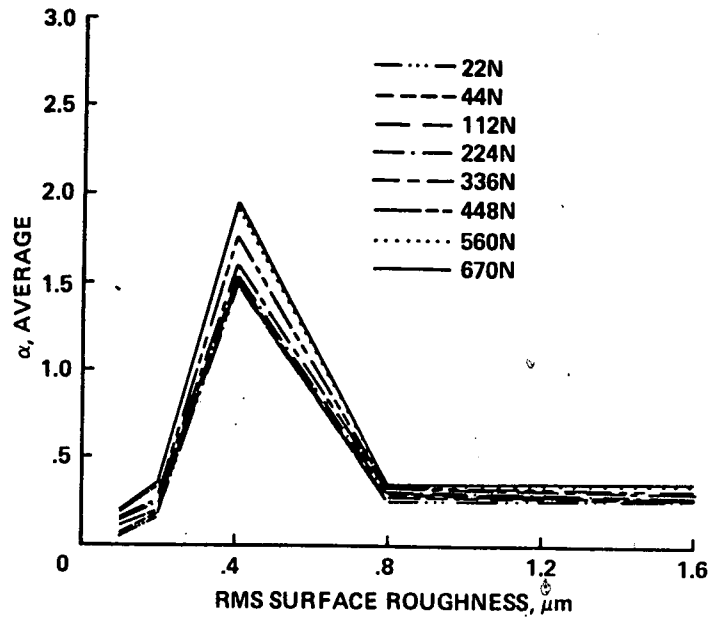


Fig. 17. Stainless steel, T = 4.2 K.

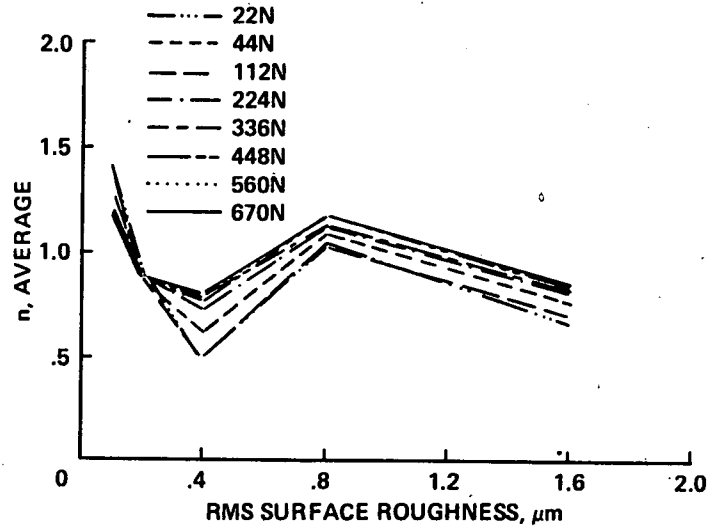


Fig. 18. Stainless steel, T = 4.2 K.

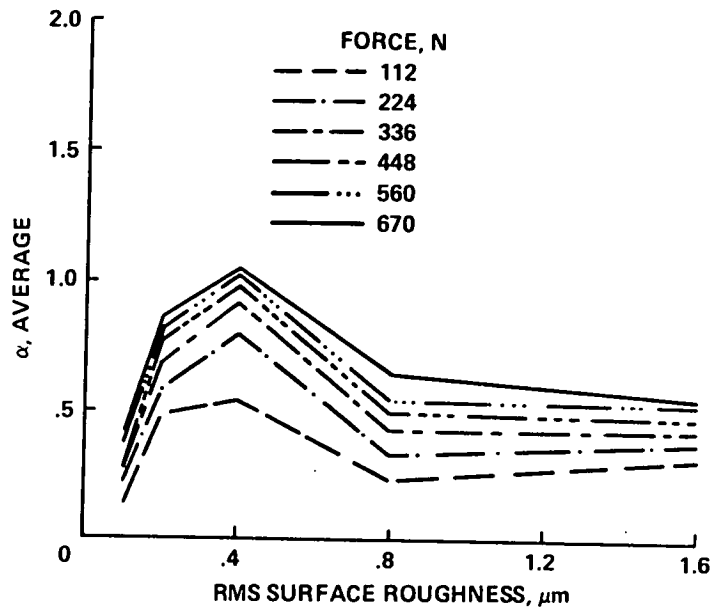


Fig. 19. Brass, $T = 4.2 \text{ K}$.

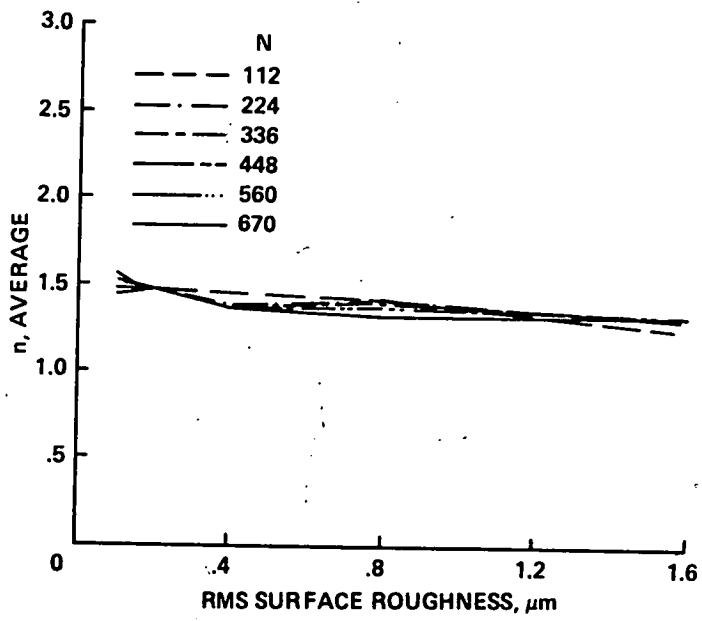


Fig. 20. Brass, $T = 4.2 \text{ K}$.

1. Report No. NASA TM 86855	2. Government Accession No.	3. Recipient's Catalog No.	
4. Title and Subtitle THERMAL CONDUCTANCE OF PRESSED ALUMINUM AND STAINLESS STEEL CONTACTS AT LIQUID HELIUM TEMPERATURES		5. Report Date February 1986	6. Performing Organization Code
		8. Performing Organization Report No. 86033	10. Work Unit No.
7. Author(s) L. J. Salerno, P. Kittel, F. E. Scherkenbach and A. L. Spivak*		11. Contract or Grant No.	
		13. Type of Report and Period Covered Technical Memorandum	
9. Performing Organization Name and Address Ames Research Center, Moffett Field, CA 94035 *Trans-Bay Electronics, Inc., Richmond, CA 94804		14. Sponsoring Agency Code 506-45-31	
		12. Sponsoring Agency Name and Address National Aeronautics and Space Administration Washington, DC, 20546	
15. Supplementary Notes Point of contact: L. T. Salerno, Ames Research Center, MS 244-10, Moffett Field, CA 94035 (415) 694-6526 or FTS 464-6526			
16. Abstract <p>The thermal conductance of aluminum and stainless steel 304 sample pairs with surface finishes ranging from 0.1 to 1.6μm rms roughness has been investigated over a temperature range from 1.6 to 6.0 K. The thermal conductance follows a simple power law function of temperature, with the exponent ranging from 0.5 to 2.25, increases asymptotically with increasing applied force, and exhibits an anomaly for surface finishes in the 0.4-μm region.</p>			
17. Key Words (Suggested by Author(s)) Thermal conductance Cryogenic instrument design Liquid Helium Pressed contacts		18. Distribution Statement Unlimited Subject category: 34	
19. Security Classif. (of this report) Uncl	20. Security Classif. (of this page) Uncl	21. No. of Pages 15	22. Price* A02