

Contract No. NAG 9-310
Accounting Code:
992-15-MU-04-6A-BL-2511-EC
309-02-01-02-6A-BL-2511-EC

FLOW BOILING WITH
ENHANCEMENT DEVICES FOR
COLD PLATE COOLANT CHANNEL DESIGN

SEMIANNUAL REPORT
February 25, 1989

Submitted to the
National Aeronautics and Space Administration (NASA)
Lyndon B. Johnson Space Center

by

Ronald D. Boyd, Sr., Ph.D.
Department of Mechanical Engineering
College of Engineering
Prairie View A&M University
Prairie View, TX 77446

Contract No. NAG 9-310
Accounting Code:
992-15-MU-04-6A-BL-2511-EC
309-02-01-02-6A-BL-2511-EC

FLOW BOILING WITH ENHANCEMENT
DEVICES FOR COLD PLATE COOLANT CHANNEL DESIGN

SEMIANNUAL REPORT
February 25, 1989

PRAIRIE VIEW A&M UNIVERSITY
COLLEGE OF ENGINEERING



Dr. Ronald D. Boyd, Sr., PI
and Professor

~~PRECEDING PAGE BLANK NOT FILMED~~

**FLOW BOILING WITH ENHANCEMENT
DEVICES FOR COLD PLATE COOLANT CHANNEL DESIGN**

Ronald D. Boyd (PI), Jerry C. Turknett, and
Alvin Smith (Graduate Students)
Department of Mechanical Engineering
Prairie View A&M University
Prairie View, TX 77446

ABSTRACT

A three-year research program has been launched to study the effect of enhancement devices on flow boiling heat transfer in coolant channels, which are heated either from the top side or uniformly. Although the study will include other orientations and working fluids in subsequent years, the first years will involve studies of the variations in the local (axial and circumferential) and mean heat transfer coefficients in horizontal, top-heated coolant channels with smooth walls and internal heat transfer enhancement devices. Initially, the working fluid will be freon-11. The objectives of this fiscal year's work are to: (1) examine the variations in both the mean and local (axial, and circumferential) heat transfer coefficients for a circular coolant channel with either smooth walls or with both a twisted tape and spiral finned walls, (2) examine the effect of channel diameter (and the length-to-diameter aspect ratio) variations for the smooth wall channel, and (3) develop an improved data reduction analysis.

This effort is intended to lead to the development of fundamentally-based heat transfer correlations which include effects of: (1) complex heat flux distributions, (2) enhancement

device configuration, and (3) basic flow parameters. This overall effort is intended to set the stage for the study of heat transfer and pressure drop in single-side heated systems under zero-gravity conditions.

The case of the top-heated, horizontal flow channel with smooth walls (1.37 cm inside diameter, and 122 cm heated length) has been completed. The data has been reduced using a preliminary analysis based on the heated hydraulic diameter. Preliminary examination of the local heat transfer coefficient variations indicates that there are significant axial and circumferential variations in the local heat transfer coefficient. However, it appears that the circumferential variation is more significant than the axial ones. Integrated averaged heat transfer coefficients will be obtained after the improved data reduction model has been implemented. In particular, this work will result in the inclusion of effects associated with circumferential variations in the effective heat flux at the inside wall.

The experimental progress has been interrupted over the last two months because the entire flow loop had to be moved from the "Old Engineering Building" (which is now being renovated) to the Agriculture Research Center. Testing is scheduled to resume in about one week.

Two graduate students (Jerry C. Turknett and Alvin Smith) will give presentations, based on the ongoing work, at the NASA-HBCU Space Science and Engineering Research Forum, in Huntsville (AL) on March 23, 1989.

CONTENTS

	Page
ABSTRACT.....	iii
NOMENCLATURE.....	vi
INTRODUCTION.....	1
LITERATURE SURVEY.....	3
EXPERIMENTAL INVESTIGATION.....	3
RESULTS.....	9
CONCLUSIONS.....	17
ACKNOWLEDGMENTS.....	18
REFERENCES.....	18

NOMENCLATURE

h	Local heat transfer coefficient, $W/m^2 \text{ } ^\circ C$
h_n	Mean heat transfer due to natural convection between the test section and the ambient, $W/m^2 \text{ } ^\circ C$
q_c	Heat loss from the test section due to convection, W/m^2
q_R	Heat loss from the test section due to radiation, W/m^2
r	Radial coordinate for the data reduction model, m
T_f	Bulk temperature of the flowing fluid, $^\circ C$
T_m	Local measured outside wall temperature of the test section, $^\circ C$
T_{sat}	Saturation temperature (316 K at 0.19 MPa for freon-11), $^\circ C$
T_a	Ambient temperature, $^\circ C$
Z, Z_i	Axial coordinate for heated portion of the test section; in Figures 7 through 10, Z_i represents Z_1 , where $Z_1 = 20.32(i-1)$, cm

Greek Letters

ϕ	Circumferential coordinate; see Figure 4 for the datum. In Figures 7 through 10, ϕ is also referred to as "Phi."
π	Half of a full rotation or 180° ; in Figures 7 through 10 π is also referred to as "Pi."

**ORIGINAL PAGE IS
OF POOR QUALITY**

INTRODUCTION

Space commercialization will require efficient heat transfer systems. The future success of many efforts will be based on our understanding of the behavior of two-phase flow boiling in both the space (zero-g or reduced-g) and earth environments. This three-year program is intended to focus on the following fundamental characteristics of experimental flow boiling heat transfer and pressure drop in the earth environment: (1) non-uniform heat flux distribution, (2) resulting local distributions of the heat transfer coefficient, (3) pressure drop and pumping power, (4) single and double enhancement devices, (5) the relative advantages of saturated and subcooled flow boiling regimes, (6) flow channel aspect ratio effects, (7) the relative effects of heat transfer enhancement techniques, and (8) correlations for mean (and local) heat transfer and pressure. Future multi-year research efforts, which will be applicable to both the earth and zero-g environments, will include basic phenomena such as: (1) the effect of orientation and Marangoni effects, (2) flow stability, and (3) identification of the threshold inertia (Froude and modified Froude numbers). Threshold inertia determination is necessary to identify when orientation (earth or reduced-g) and/or Marangoni (at zero-g) effects become important. Although not apparent, the development of improved data reduction models are essential to the accurate representation and interpretation of the heat transfer data.

The objective of the first year's efforts are to:

(1) examine the variations in both the mean and local (axial, and circumferential) heat transfer coefficients for a circular coolant channel with either smooth walls or with both a twisted tape and spiral finned walls, (2) examine the effect of channel diameter (and length-to-diameter aspect ratio) variations for the smooth wall channel, and (3) develop an improved data reduction analysis.

Thus far, freon-11 flow boiling experiments have been completed for the case of a smooth-wall horizontal coolant channel (1.37 cm inside diameter, and 122.0 cm heated length) heated from the top side. The inlet freon-11 temperature was 24°C, the absolute exit pressure was 0.19 MPa, and the mass velocity was 0.28 Mg/m²s. Work is proceeding on: (1) heat transfer enhancements due to internal devices, (2) the examination of the effect of diameter for smooth wall channels, and (3) the development of an improved data reduction technique.

Some progress has been made to examine whether improvements are needed in the present data reduction analyses. Four models have been developed and formulated (e.g., [1]), including a preliminary finite difference model. Since all equations for each model must be verified before being used to reduce data, work is proceeding to "bench-mark" the developed models.

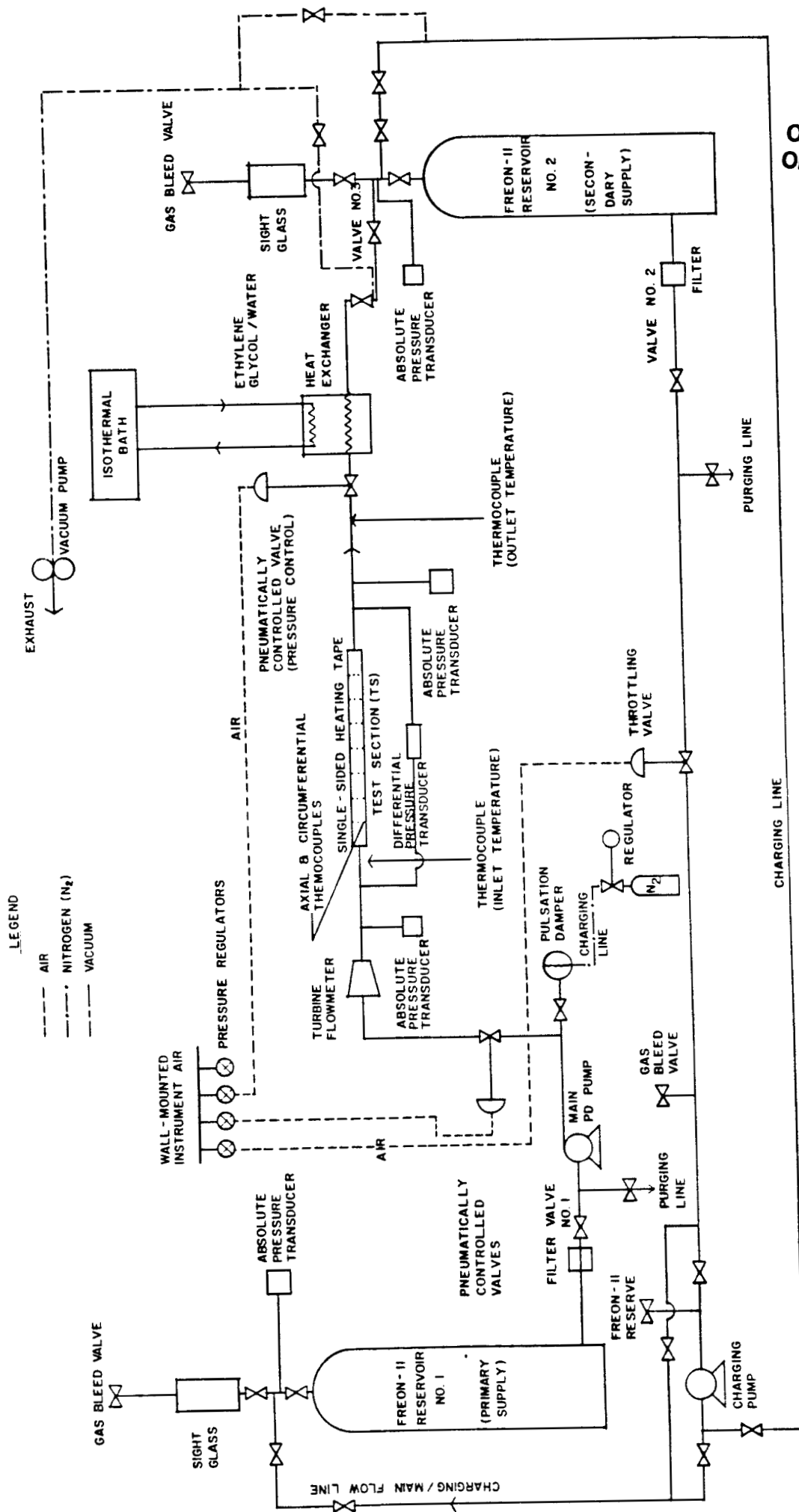
Two graduate students (Jerry C. Turknett and Alvin Smith) will give presentations at the NASA-HBCU Space Science and Engineering Research Forum, in Huntsville (AL) on March 23, 1989.

LITERATURE SURVEY

A survey of the literature is ongoing. Brief surveys of the related literature can be found in references [1 and 2].

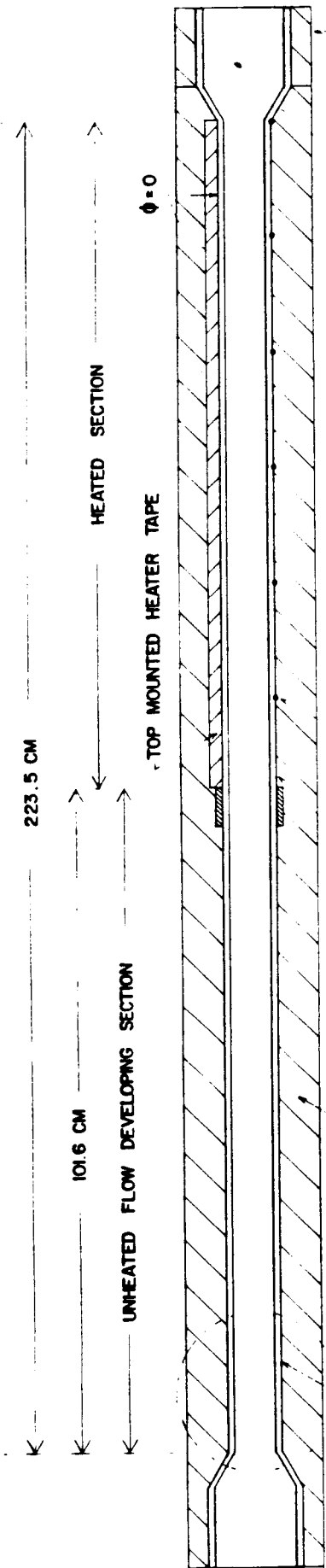
EXPERIMENTAL INVESTIGATION

The reader is referred to references [1 and 2] for detail descriptions of the experimental flow loop, procedures, and data acquisition. Figures 1 and 2 show both the flow loop and test section configurations, respectively. Figure 3 displays the heated portion of the test section at a stage of the assembly where the axial and circumferential placements of the thermocouples (wires hanging below the horizontal tube), as well as the top-mounted heater tape, are clearly seen. The thermocouples are used to make temperature measurements of the outside wall of the heated coolant channel. Figure 4 is a schematic of the cross section of the heated portion of the test section [which is preceded by an upstream unheated portion for flow development]. The measured wall temperatures are used along with the data reduction analysis to determine the unknown heat transfer coefficient, h . Recently, a data reduction technique based on the heated hydraulic diameter [2] (see Figure 5), was used to reduce the experimental data. This approach will result in, at most, a qualitative indication of the local distribution of h . Work is proceeding on more viable approaches; e.g., finite difference for local h and analytical for mean h .



ORIGINAL PAGE IS OF POOR QUALITY

Figure 1: Freon-11 Flow Loop for Both Subcooled and Saturated Flow Boiling Experiments.



INSULATION

SMOOTH TUBE

THERMOCOUPLES MOUNTED CIRCUMFERENTIALLY = $\pi/4, \pi/2, 3\pi/2, 7\pi/4$ AND AT SIX AXIAL LOCATIONS AT 20-32 CM INTERVALS.

DOWNSTREAM PART OF SECTION WITH OR WITHOUT ENHANCEMENT DEVICE

REMOVABLE INSULATION WHEN CHANGING TEST SECTION

ORIGINAL PAGE IS OF POOR QUALITY



Figure 2: Horizontal Test Section for Measuring Local Heat Transfer Coefficient Distributions.

ORIGINAL PAGE IS
OF POOR QUALITY

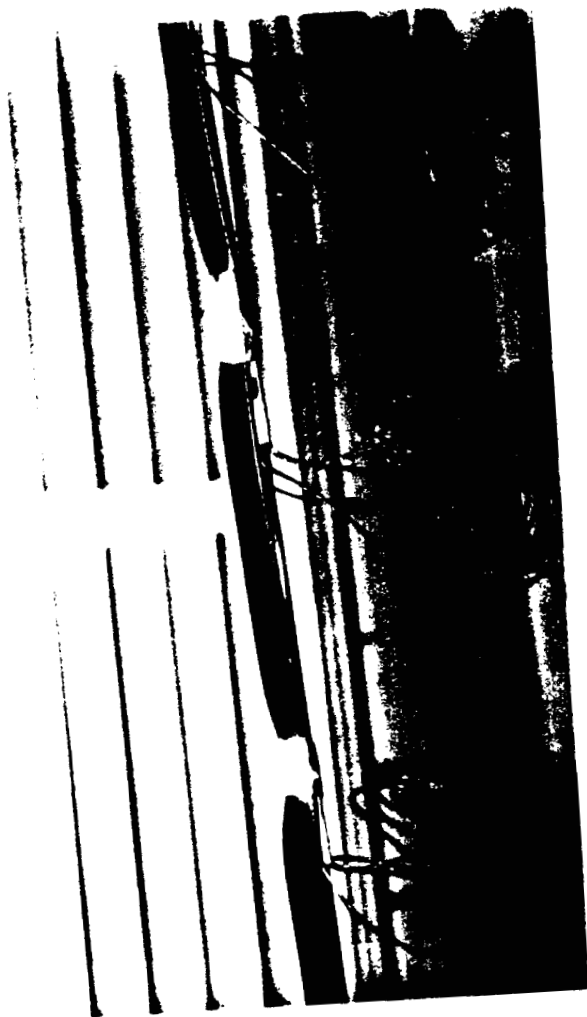


Figure 3: Thermocouple and Heater Tape Installation.

ORIGINAL PAGE IS
OF POOR QUALITY

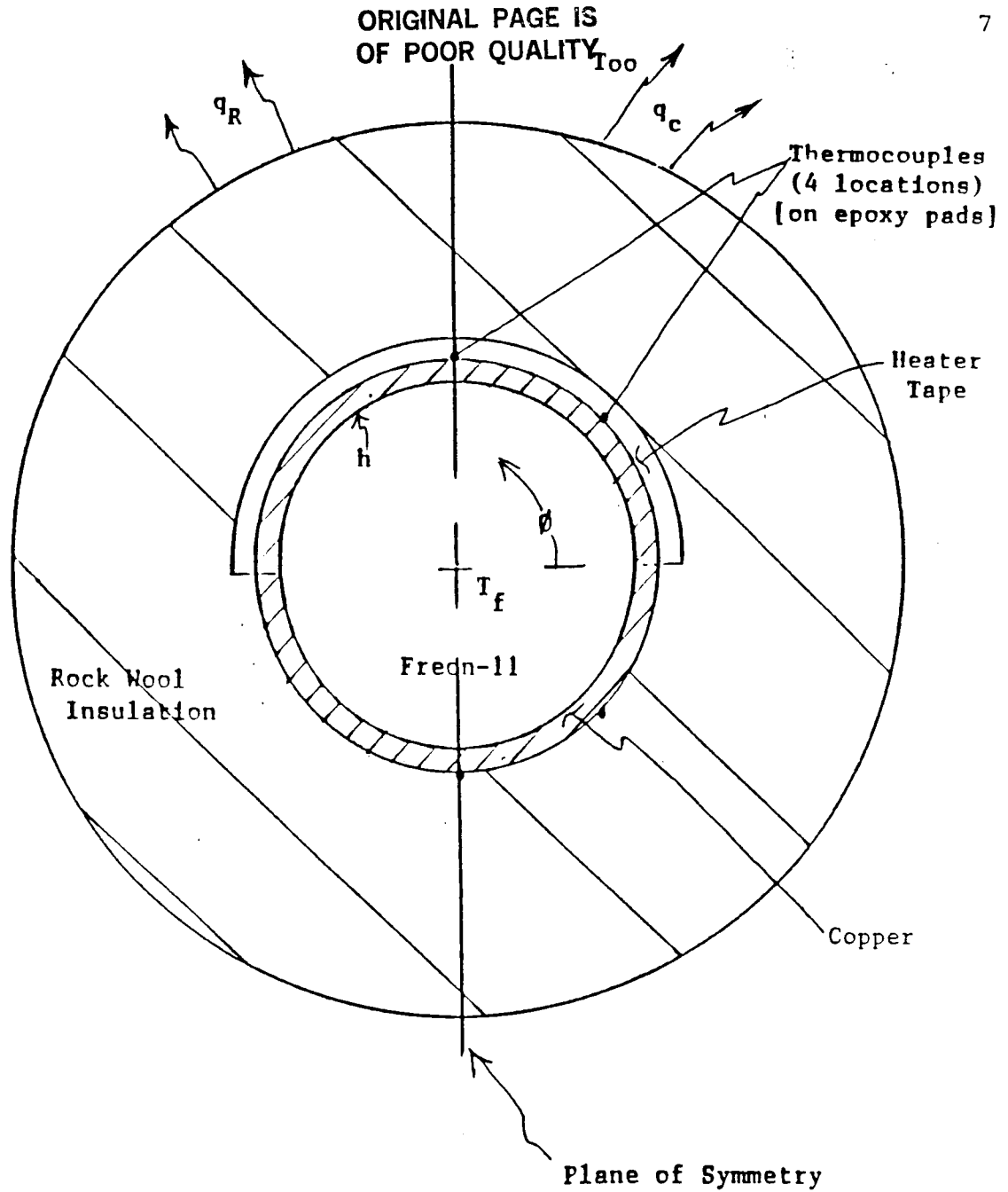


Figure 4: Cross Section of the Heated Portion of the Test Section.

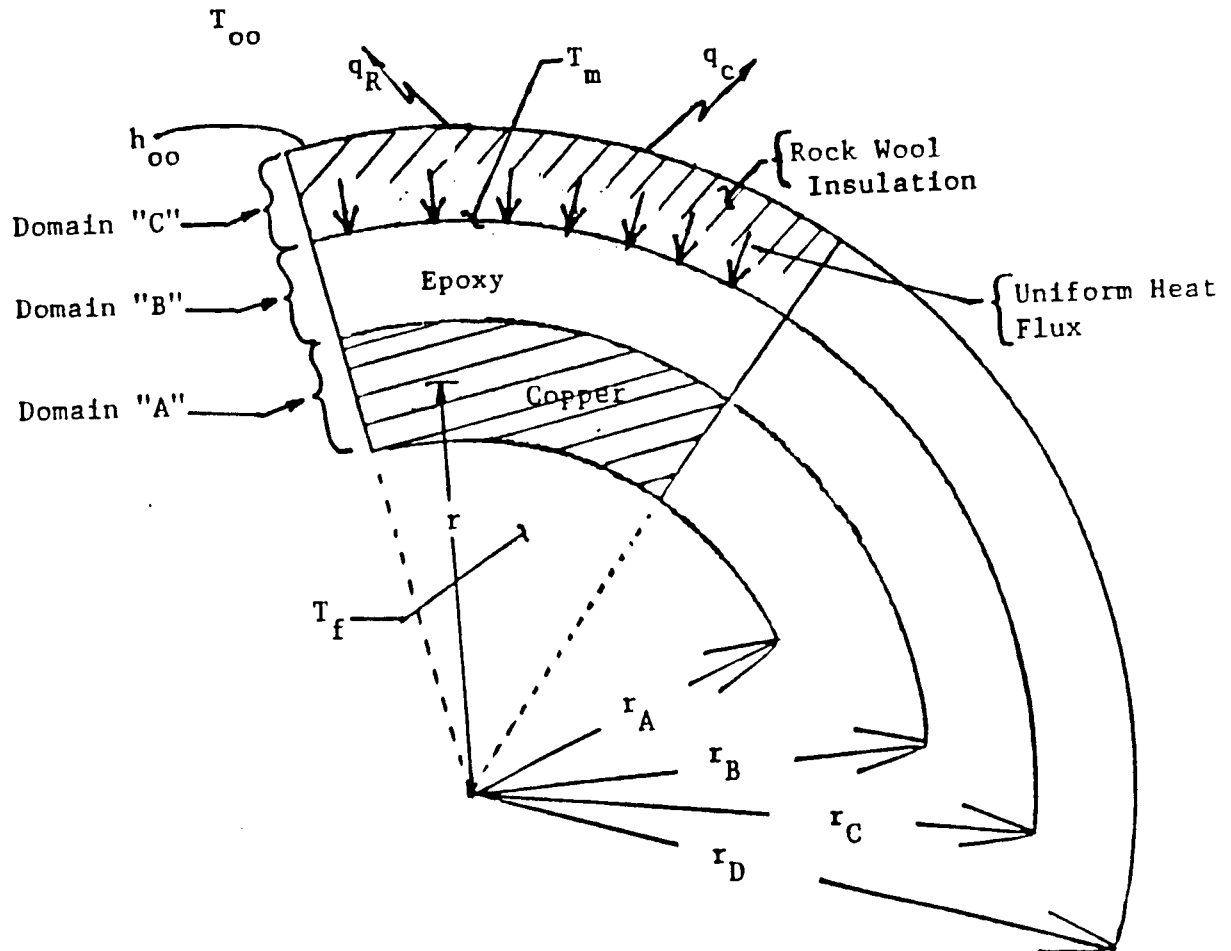


Figure 5: Control Volume for the Heated Hydraulic Diameter Model.

In applying either model, knowledge of the fluid's bulk temperature must be used. An iteration scheme is used to compute the inside wall temperature, T_w , of the flow channel. The fluid's temperature is chosen based on the magnitude of the inside wall temperature relative to the wall temperature required to cause the onset to nucleate boiling (T_{ONB}). If T_w is greater than T_{ONB} , the fluid temperature is set equal to the saturation temperature. However, if the above condition is not satisfied, the fluid temperature is computed from the energy equation, using the measured inlet fluid temperature and the measured net thermal energy transfer to the fluid.

RESULTS

The results are presented for the case of a horizontal flow channel with smooth walls (1.37 cm inside diameter, and 122 cm heated length) which is heated from the top side. The flow rate and exit pressure of the freon-11 were maintained at $0.281 \text{ Mg/m}^2\text{s}$ and 0.19 MPa (316 K saturation temperature), respectively. The inlet freon temperature was held constant at 24.0°C .

The variation in the heat transfer coefficient is more pronounced in the circumferential direction than the axial direction. The heat transfer coefficients, for the single-phase and hypocrritical regions [3] are shown in Figure 6. The fully developed nucleate boiling region does not appear in the figure but occurs somewhere in a narrow range between $\phi = 7\pi/4$ and $\pi/4$

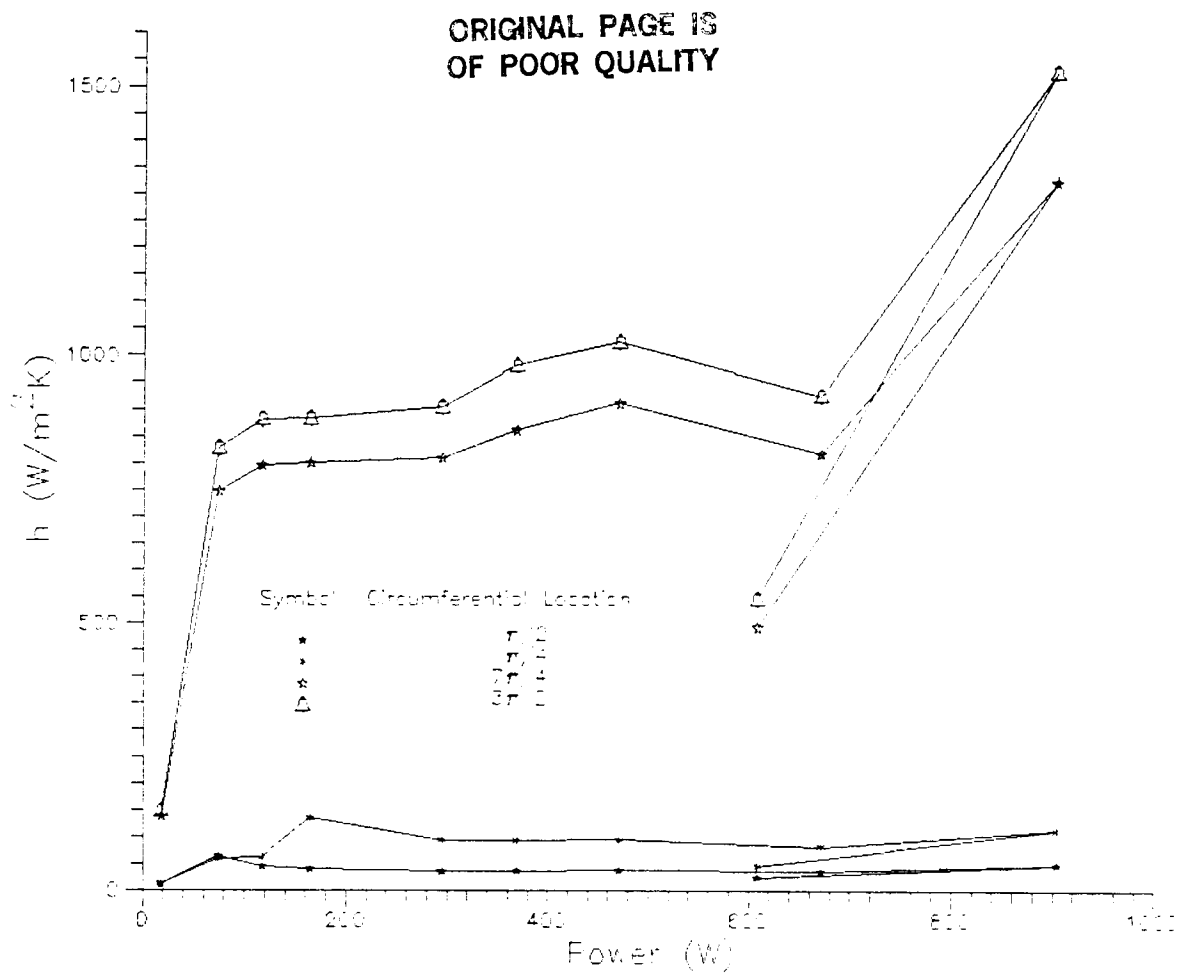


Figure 6: Heat Transfer Coefficient Versus Power Generation and Circumferential Location at $Z = Z_4 = 61.96$ cm (center of the test section) for Top-Side-Heated Smooth Tubes for: 0.19 MPa Exit Pressure, $0.281 \text{ Mg/m}^2\text{s}$ mass velocity, 1.22 m Heated Length.

ORIGINAL PAGE IS
OF POOR QUALITY

(see Figure 4 for the datum for ϕ). It is expected that the heat transfer coefficient in this narrow region to be greater than any of those shown. It should be noted that the catastrophic drop in h for each of the curves is due to the heat tape becoming overheated and as a result eventually malfunctioning.

The last two highest points on each of the curves show an increase in the heat transfer coefficient. For $\phi = 3\pi/2$ and $7\pi/4$, the increase is due to the onset of nucleate boiling which we see results in more than a fifty percent increase in h . For these two locations we see that there is only a secondary variation in h with ϕ . In the hypocritical region, the increase in h appears to be very minor. However, the actual increase in h is much larger than that shown on the figure because T_{sat} was used for T_r rather than its actual value. Hence, if further work is conducted in the hypocritical regime, better estimates of the actual (local) fluid temperature must be made.

Both axial and circumferential variations in h are found to be significant. Comparisons of Figures 7 ($\phi = \pi/2$), 8 ($\phi = \pi/4$), 9 ($\phi = 7\pi/4$) and 10 ($\phi = 3\pi/2$) reveal the complex nature of the variations. The variation in the local heat transfer coefficient increases from the bottom ($\phi = 3\pi/2$) to the top ($\phi = \pi/2$) of the test section at all axial locations. As noted earlier, the variation between $7\pi/4$ and $3\pi/2$ is small even at the locations where incipient nucleate boiling occurs ($h = 1400 \text{ W/m}^2\text{K}$). It is interesting to compare the magnitude of h for the three regimes:

ORIGINAL PAGE IS
OF POOR QUALITY

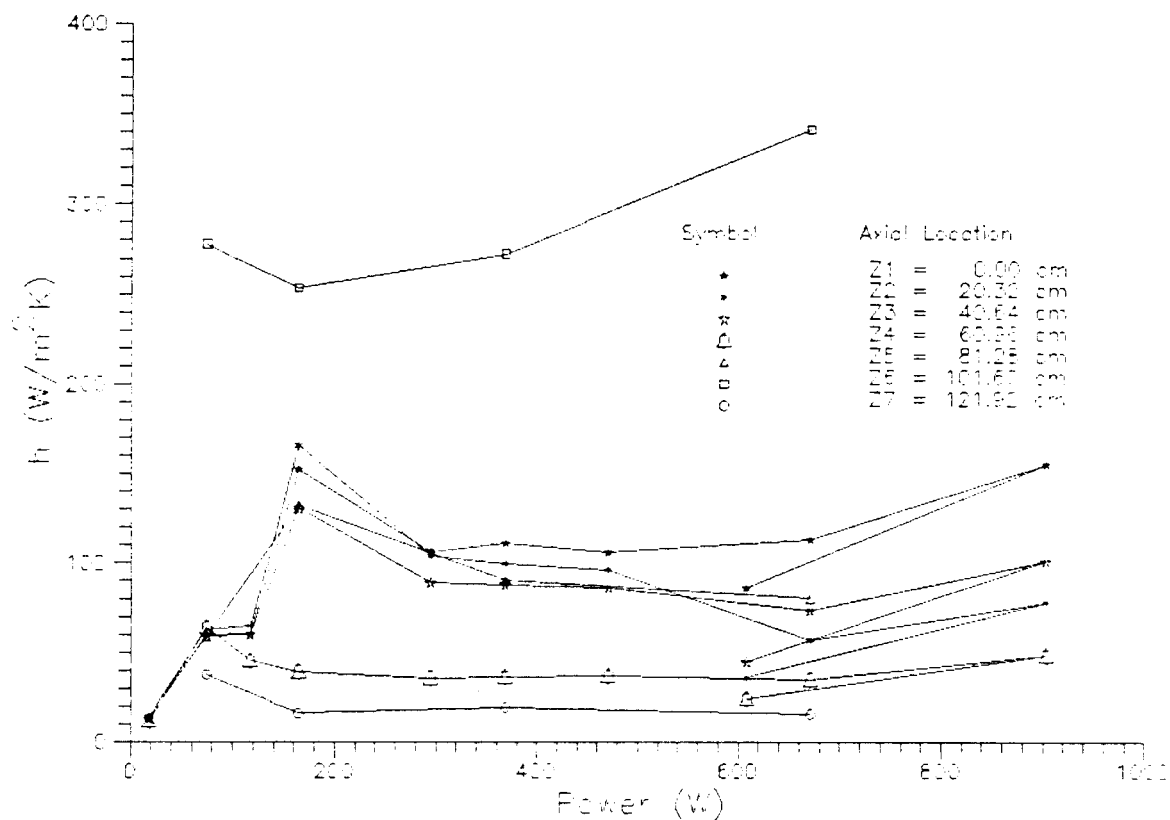


Figure 7: Heat Transfer Coefficient Versus Power Generation and Axial Location at $\Phi = \pi/2$ (top of the test section) for Top-Side-Heated Smooth Tubes for: 0.19 MPa Exit Pressure, $0.281 \text{ Mg/m}^2\text{s}$ mass velocity, 1.37 cm Channel inside Diameter, and 1.22 m Heated Length.

ORIGINAL PAGE IS
OF POOR QUALITY

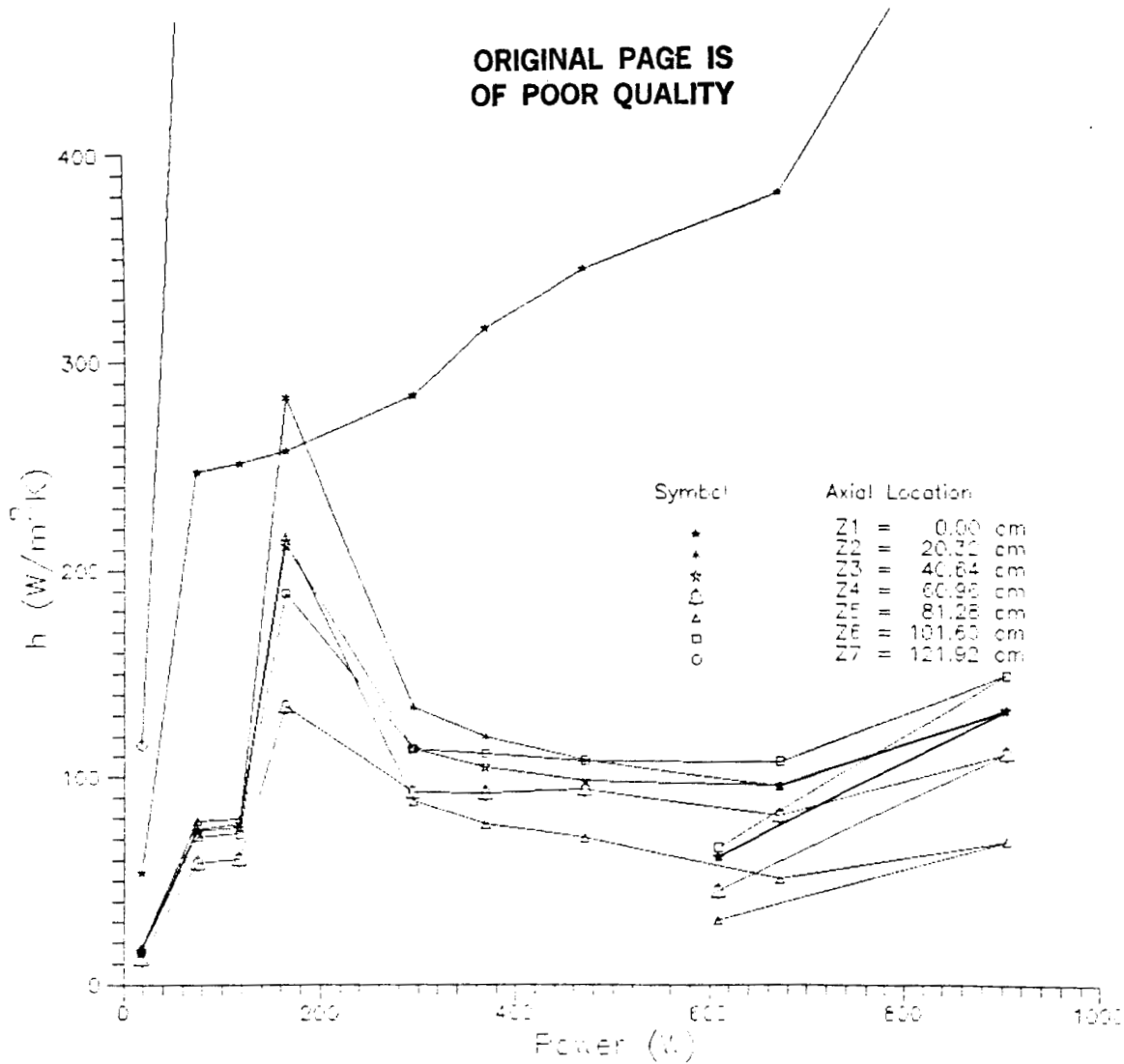


Figure 8: Heat Transfer Coefficient Versus Power Generation and Axial Location at $\Phi = \pi/4$ for Top-Side-Heated Smooth Tubes for: 0.19 MPa Exit Pressure, 0.281 Mg/m^2 mass velocity, 1.37 cm Channel inside Diameter, and 1.22 m Heated Length.

ORIGINAL PAGE IS
OF POOR QUALITY

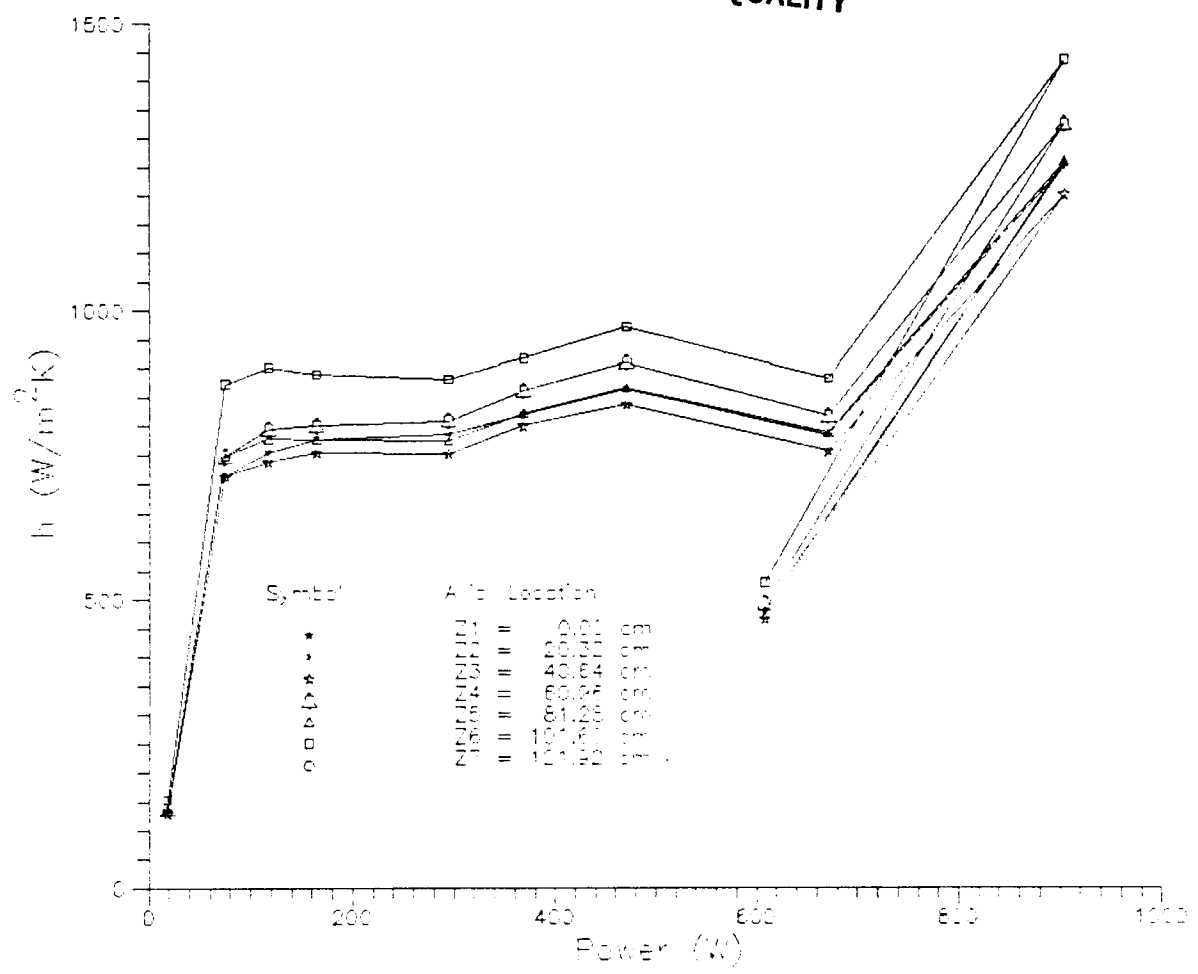


Figure 9: Heat Transfer Coefficient Versus Power Generation and Axial Location at $\Phi = 7\pi/4$ for Top-Side-Heated Smooth Tubes for: 0.19 MPa Exit Pressure, $0.281 \text{ Mg/m}^2\text{s}$ mass velocity, 1.37 cm Channel inside Diameter, and 1.22 m Heated Length.

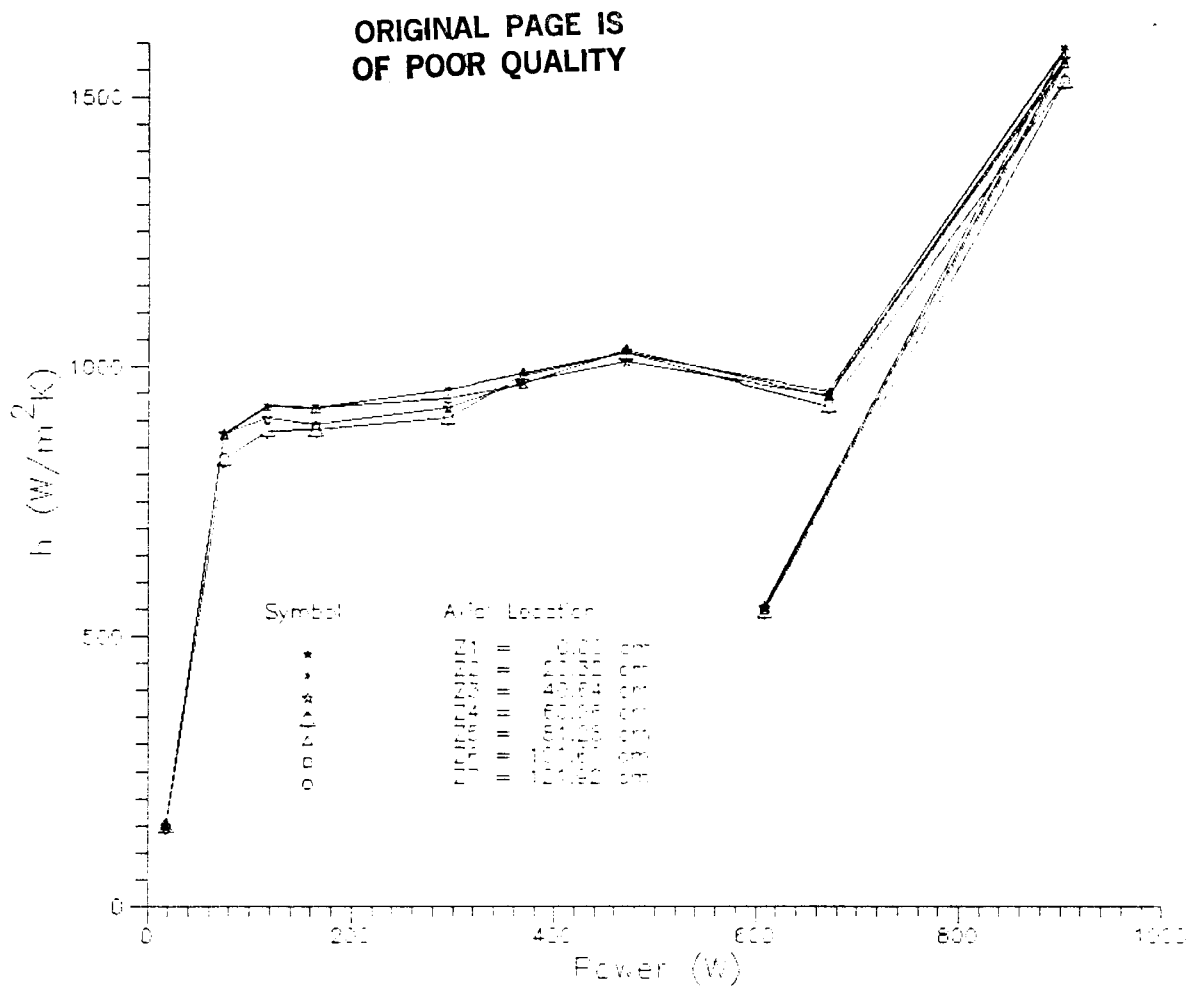


Figure 10: Heat Transfer Coefficient Versus Power Generation and Axial Location at $\Phi = 3\pi/2$ (bottom of the test section) for Top-Side-Heated Smooth Tubes for: 0.19 MPa Exit Pressure, 0.281 Mg/m²s mass velocity, 1.37 cm Channel inside Diameter, and 1.22 m Heated Length.

(1) single-phase ($800 \text{ W/m}^2\text{K}$), (2) incipient nucleate boiling ($1400 \text{ W/m}^2\text{K}$) and, (3) film boiling (10 to $100 \text{ W/m}^2\text{K}$).

If one takes time to study the relative positions of the curves and use the reduced wall temperature, some of the character of the flow is revealed (e.g., See Figure 7). In particular, notice from Figure 7 that: (1) h at Z_5 and Z_6 are almost identical at between 380 and 640 W , (2) h at Z_4 is much higher than all values of h at other locations, and (3) in some cases curves are crossing one another. These observations may imply a slug type flow. For example, at Z_4 the unusually large value of the heat transfer coefficient could be due to local cooling (slug flow). Contrasting the above description, Figure 8 ($\phi = \pi/4$) shows that between Z_2 and Z_3 the heat transfer coefficient decreases in the downstream direction. This is consistent with the previous observations made. That is, at $\phi = \pi/4$ the film boiling regime predominates.

CONCLUSIONS

Local (axial and circumferential) measurements of the outside wall temperature have been made for horizontal freon-11 flow (0.19 MPa exit pressure and 24°C inlet temperature) through a 1.37 cm inside diameter coolant channel with smooth walls and heated from the top side. A preliminary data reduction model was used to relate the measured wall temperatures to the local heat transfer coefficient. Although the local temperature measurements are quantitative, the preliminary data reduction model results in what may be only qualitative local heat transfer measurements. Work is proceeding to evaluate and improve, if necessary, the existing data reduction model.

The preliminary heat transfer data indicates that there are significant axial and circumferential variations in the local heat transfer coefficient. However, it appears that the circumferential variation is more significant than the axial ones. The single phase heat transfer coefficient (near 900 W/m²K) is increased by more than 50% at the onset of nucleate boiling. For the test performed, the circumferential heat transfer coefficient varied from the hypocrritical to the single phase heat transfer regimes. This resulted [in some cases] in a factor of ten increase in the local heat transfer coefficient. The axial variations rarely exceeded a factor of three.

ACKNOWLEDGMENTS

The authors would like to acknowledge Richard Parish, Dr. Joseph Atkinson, Dr. Y. Freeman, and Richard Kennedy for their support and assistance. The authors would also like to thank Mr. Jim Gips for his assistance in finalizing details for the formal placement of the contract. Finally, the authors are appreciative to other NASA personnel, both at headquarters and JSC, for their support.

REFERENCES

1. Boyd, R. D., "Flow Boiling with and without Enhancement Devices for Horizontal, Top-Heated, Coolant Channels for Cold Plate Design Applications, Preliminary Report," Department of Mechanical Engineering, Prairie View A&M University, Prairie View, TX, Submitted to NASA (JSC), Contract no. 9-16899 (Task-5), December, 1986.
2. Turknett, J. C., "Forced Convection and Flow Boiling with and without Enhancement Devices for Top-Side-Heated Horizontal Channels," MS Thesis, Department of Mechanical Engineering, Prairie View A&M University, Prairie View, TX (to be published).
3. Cumo, M., and Naviglio, A., Thermal Hydraulics, CRC Press, Inc., Boca Raton, Florida, 1988, II, pp 59-68.

DISTRIBUTION

Mr. Richard Parish (3)
NASA (JSC)
Mail Code FC-2
Houston, TX 77058

NASA Scientific & Information Facilities (2)
P. O. Box 8757
Baltimore-Washington Airport
Baltimore, MD 21240

Prairie View A&M Research Foundation
P. O. Box 3578
Attention: Robert Effinger

Dr. Wayne D. Perry, Dean
College of Engineering
Prairie View A&M University
Prairie View, TX 77446

Dr. Shield Lin, Chairman
Department of Mechanical Engineering
Prairie View A&M University
Prairie View, TX 77446

Dr. Ronald D. Boyd, Sr.
Department of Mechanical Engineering
Prairie View A&M University
Prairie View, TX 77446