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Performance of All-Metal Demountable Cryogenic Seals at Superfluid Helium Temperatures

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PERFORMANCE OF ALL-METAL DEMOUNTABLE CRYOGENIC SEALS
AT SUPERFLUID HELIUM TEMPERATURES*

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Abstract

Two all-metal demountable cryogenic seals with an outside diameter of 36.6 mm, inside diameter of 27.2 mm, and thickness of 0.51 mm were leak-tested at room temperature (300 K), liquid nitrogen temperature (21 cycles at 77 K), liquid helium temperature (9 cycles at 4.2 K), and superfluid helium temperature (4 cycles at 1.6 K). Each seal was mounted and demounted for 13 cycles. Thickness measurements at 90° intervals along the circumference showed a maximum seal compression of 0.038 mm. Leak-rate measurements at all temperatures showed no detectable leak above the helium background level, typically 0.1×10^{-9} std-cc/sec, during testing.

I. Introduction

In the past, Wood's metal, copper, aluminum, tin, and indium have been used to make demountable seals for cryogenic applications. Indium in particular has been used with excellent results because of the ease of fabricating an O-ring from indium wire. One of the problems of using indium is that leakage frequently occurs under rapid cooling as a result

of the differential contraction between the seal and the mating materials.¹

Such commercial products as Varian Conflat™ flanges have been used to address the sealing problem by providing a disposable copper ring which seals against stainless steel knife edges. These seals provide excellent results. The only disadvantage is that a new gasket must be used each time the flange is disassembled and reassembled. A need exists for a demountable cryogenic seal which is leak-tight at superfluid helium (SfHe) temperatures after repeated cycling.

Creare, Inc., developed a reusable static cryogenic seal for applications such as cryogenic turbo-expander inlet and exhaust lines, and cryogenic sensor mounts.²

The annular-shaped seal consists of a flat indium/silver-alloy seal surrounded at both the inside and outside diameters by a 316 stainless steel guard. The seal has an outside diameter of 36.6 mm, an inside diameter of 27.2 mm, and a thickness of 0.51 mm (see Fig. 1).

*The identification in this paper of certain commercial products does not imply recommendation or endorsement by the National Aeronautics and Space Administration, nor does it imply that the products identified are necessarily the best available for the purpose.

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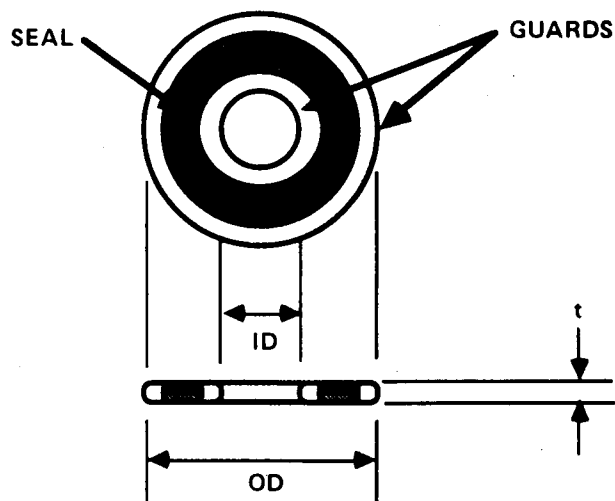


Fig. 1 Diagram of seal.

The manufacturer recommends face mounting between bolted flanges with a maximum of five complete mounting/demounting cycles. Flange bolts are to be torqued in at least three steps to generate a compressive force of 11,000 lbf (49,000 N) \pm 110 lbf (490 N) on the seal.

II. Method

Figures 2 and 3 show the flange assembly designed for testing the seals. The smooth surfaces of two Varian

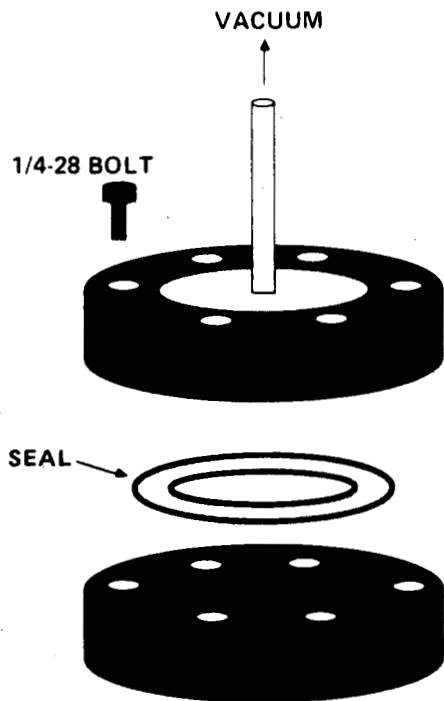


Fig. 2 Conflat™ flange assembly.



Fig. 3 Flange assembly.

Conflat™ flanges were used as the sealing surfaces. A port drilled in the upper flange accommodated a 0.25-in. (6.3-mm) evacuation tube which was silver soldered to the flange. Figure 4 shows one of the seals resting on the lower flange. The upper flange is to the right and the evacuation port is visible in the center of the flange.

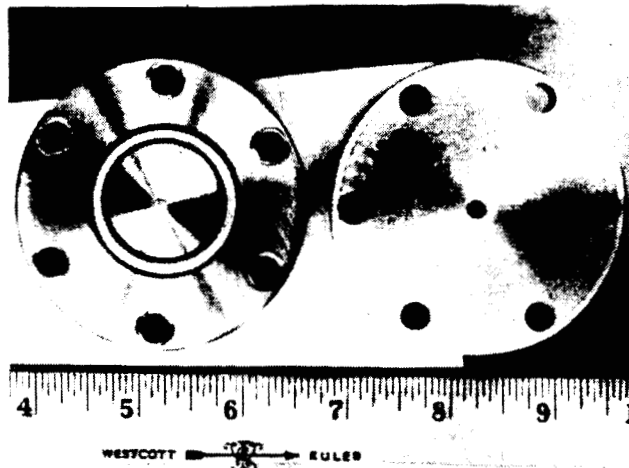


Fig. 4 Seal resting on flange.

The flanges were bolted together with six 1/4-28 bolts and nuts. The tightening torque required to generate a compressive force of 11,000 lbf (49,000 N) at the seal was calculated to be 91.5 in-lb (10.4 N-m) per bolt.

The flange assembly was mounted into a cryostat and installed in a Dewar. By evacuating the Dewar, SHe temperatures were achieved. Figure 5 shows the experimental setup. Table 1 presents the details of the test procedure.

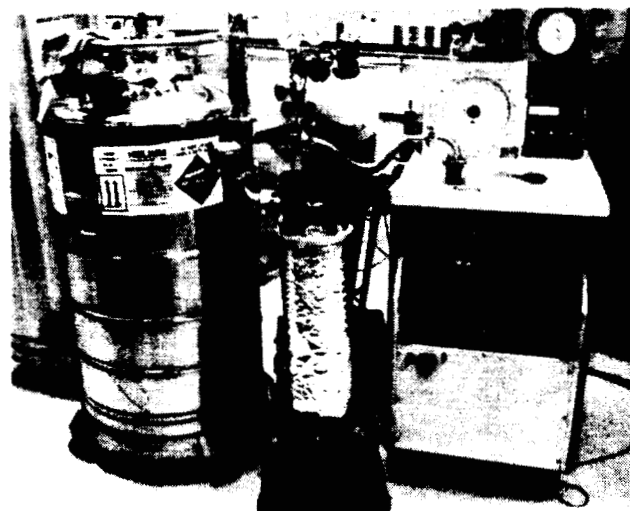


Fig. 5 Experimental setup.

Table 1 Cryogenic seal test procedure

Task	Nominal leak check temperature			
	300 K	77 K	4.2 K	1.6 K
Assembly	x			
LN ₂ leak test	x	x		
Room-temperature leak test	x			
LHe leak test (5 cycles)	x		x	
Room-temperature leak test	x			
Cryostat mounting	x			
SfHe leak test	x	x	x	x
Room-temperature leak test	x			
Disassembly				
Seal inspection/measurement				
Reassembly	x			
LN ₂ leak test	x	x		
Room-temperature leak test	x			
Cryostat mounting	x			
SfHe leak-test procedure	x	x	x	x
Room-temperature leak test	x			
Disassembly				
Seal inspection/measurement				
Reassembly	x			
LN ₂ leak test (7 cycles)	x	x		
Room-temperature leak test	x			
Disassembly				
Seal inspection/measurement				
Reassembly	x			
Cryostat mounting	x			
SfHe leak test	x	x	x	x
Room-temperature leak test	x			
Disassembly				
Seal inspection/measurement				
Reassembly	x			
LN ₂ leak test (6 cycles)	x	x		
Room-temperature leak test	x			
Disassembly				
Seal inspection/measurement				
Reassembly	x			
Cryostat mounting	x			
SfHe leak test	x	x	x	x
Room-temperature leak test	x			

Prior to testing, the seal was inspected for pits, irregularities, and radial scratches on the stainless steel guards, which could affect the sealing properties. Grooves were noted on the seals in the circumferential direction. Small nicks and scratches, which proved to be inconsequential, were present on the indium/silver surface. Measurements of

the seal thickness were taken at 90° locations around the circumference of the seal.

The seal was centered between the flanges, and the bolts were installed. A tightening sequence was used to assure a uniform preload of the seal. The sequence pattern was repeated in four steps beginning with a torque of 25 in-lb

(2.84 N-m) and with increases of approximately 25 in-lb per step. This procedure was followed wherever reassembly was appropriate.

III. Results

During the tests, each seal underwent 21 temperature cycles at liquid nitrogen (LN₂) temperature, 9 cycles at liquid helium (LHe) temperature, and 4 cycles at SfHe temperature. In addition, each seal was mounted and demounted 13 times.

For the first seal tested, the measured leak rate was below the background at all times during the tests. Background was typically 0.1×10^{-9} std-cc/sec, reaching a maximum of 0.4×10^{-9} std-cc/sec within a short time after filling the Dewar with LHe, because of the large amount of helium gas in the laboratory.

Results of the thickness measurements showed that, starting with a measured thickness of 0.0200 in. (0.51 mm), minimum thickness after 15 assembly/disassembly cycles was 0.0185 in. (0.470 mm) at one of the 90° quadrants. The other three values were 0.0200, 0.0195, and 0.0190 in.

For the second seal tested, the procedure of filling the Dewar was altered to allow for filling it outside the laboratory. This kept the residual background at a maximum of 0.1×10^{-9} std-cc/sec during helium and SfHe testing. As with the first seal tested, the measured leak rate was below the background at all times during the tests.

Results of the thickness measurements showed that, starting with a measured thickness of 0.0200 in. (0.51 mm), minimum thickness after 15 assembly/disassembly cycles was 0.0190 in. at all four quadrants. Prior to testing, the seal was marked on both sides of the stainless steel guard, and at each assembly/disassembly the seal was replaced with the two sides exchanged from the previous position.

IV. Discussion

The results indicate that for both seals tested no measurable leak occurred either at LN₂, LHe, or SfHe temperatures, even after the flange assembly was valved off from the leak detector at one point for 10 minutes at SfHe temperature to determine any accumulation effects.

Cooldown and warmup were rapid. Cooldown was accomplished by immersing the flange assembly in cryogen and warmup was accomplished by immersing the flange in hot water. Ice formation, as a result of transferring the flange assembly from cryogen to hot water, did not appear to affect the seals.

After the test was completed, some dirt was present on the indium/silver sealing surface in a few places around the seals' circumference. This did not appear to affect the sealing properties, and the dirt was easily removed with a cloth moistened in ethanol.

Aside from the slight compression of the seal noted in the results, no degradation of seal performance resulting from temperature cycling and/or mounting/demounting were noted, even though the number of mounting/demounting cycles exceeded the manufacturer's recommendation by a factor of 2.6.

V. Conclusion

The experimental results of testing two all-metal, demountable, cryogenic seals showed that after 13 complete mounting and demounting cycles and 21 cryogenic temperature cycles, 9 of which were at 4.2 K and 4 of which were at 1.6 K, the measured leak rate of the seals was below the residual background of the leak detector. This background was typically 0.1×10^{-9} std-cc/sec and reached a maximum of 0.4×10^{-9} std-cc/sec because of the large quantities of helium transfer gas present in the laboratory.

These results exceed the manufacturer's data by an order of magnitude for the leak rate and a factor of 2.6 for the mounting/demounting recommendation.

VI. References

¹White, G. K., "Experimental Techniques in Low-Temperature Physics," Clarendon Press, Oxford, 1979.

²U-Guard Seal, Creare, Inc., Etna Road, P.O. Box 71, Hanover, NH 03755.



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