# NASA Technical Memorandum 101644

# Simulated Dynamic Response of a Servovalve Controlled Hydraulic Actuator

D. A. Babcock

June 1990

(NASA-TM-101044) SIMULATED DYNAMIC RESPUNSE DE A GERVOVALVE CONTROLLED HYDRAULIC ACTUATUR (NASA) 30 D CSCL 131 N90-25161

03/31 0292967



Langley Research Center Hampton, Virginia 23665-5225

	 <del></del>	
		+
		,
		,
		•

# Simulated Dynamic Response of a Servovalve Controlled Hydraulic Actuator

## By: D. A. Babcock

#### Abstract

A general purpose math model of a servovalve controlled hydraulic actuator system is derived. The system consists of a linear actuator with unequal piston areas, a single stage servovalve, a gas charged hydraulic accumulator, and the interconnecting piping. The state equations are integrated using the Advanced Continuous Simulation Language (ACSL) for determining the system's dynamic response characteristics. Using this generalized hydraulic actuator system model, response characteristics were determined for various servovalve commands.

#### Nomenclature

#### Variables

A area

B fluid bulk modulus

C viscous damping

DP differential pressure

f servovalve bandwidth

Fst static force

Fstk sticking force

Fcol coulomb friction force

K pressure loss coefficient

m fluid mass

M total mass of piston and load

P pressure

Q volumetric flow

S laplace operator

t time

V volume

 $W_{\rm N}$  frequency

x actuator displacement

y servovalve spool displacement

 $\gamma$  ratio specific heats

au servovalve time constant

 $\rho$  fluid density

 $\xi$  damping

Subscripts

acc accumulator

act actuator

c actuator cap side

g gaseous

r actuator rod side

cmd command

in initial condition

l leakage

lq liquid

rt return side

sp supply side

tot total

vrtservovalve return sidevspservovalve supply sidevservovalvevaservovalve port avbservovalve port bvcservovalve cap sidevrservovalve rod side

#### 1.0 Introduction

Many dynamic systems use linear or rotary hydraulic actuators to generate large forces for obtaining high speed response. To satisfy design requirements for such systems, it is necessary to accurately predict load position, velocity, acceleration and system stability.

A generalized hydraulic actuator system was modeled utilizing the Advanced Continuous Simulation Language (ACSL) computer code to solve the system of nonlinear time-dependent ordinary differential equations. Model parameters are easily changed for purposes of component and system evaluation. Control strategies for particular hardware components may be evaluated by assessing the effect on system performance. For a given operating condition and controller gain setting, ACSL evaluates the poles of the system transfer function, thus providing a method to assess the stability of the system.

Three simulations were performed using typical system parameters; a step command to the servovalve, a steep ramp command, and a shallow ramp command. Results of the simulations are discussed with selected variables plotted.

## 2.0 Governing Equations

Shown in Figure 1 is a schematic of a typical servovalve controlled hydraulic actuator system. The system model consists of an actuator, single stage servovalve, gas charged

accumulator, and interconnecting piping. Presented here is the development of the governing equations for each component. The model of the hydraulic system presented here is an extension of that presented in reference (1) and in addition includes an unequal area piston actuator, Coulomb friction, sticking and pressure losses in the piping system.

#### 2.1 Actuator

The actuator is modeled as a second order spring-mass system with viscous damping, Coulomb friction and sticking. The cap and rod side piston areas are unequal. Piston seal leakage is assumed laminar and given by:

$$Q_l = K_{act}(P_c - P_r) \tag{1}$$

Referring to Figure 1, continuity of mass flow into the cap end of the actuator yeilds:

$$\frac{dm_c}{dt} + Q_l \rho_l = Qvc\rho_c \tag{2}$$

The mass contained in the cap end is given by:

$$m_c = \rho_c V_c \tag{3}$$

Differentiating,

$$\frac{dmc}{dt} = \rho_c \frac{dVc}{dt} + Vc \frac{d\rho_c}{dt} \tag{4}$$

Fluid bulk modulas is defined as:

$$B \equiv \frac{dP}{d\rho/\rho} \tag{5}$$

Volume rate of change is given as:

$$\frac{dVc}{dt} = Ac\frac{dx}{dt} \tag{6}$$

Substitute (4), (5), and (6) into (2) yeilding:

$$Qvc = \frac{Vc}{B}\frac{dPc}{dt} + Ac\frac{dx}{dt} + Q_l \tag{7}$$

Similarly for the rod end:

$$Qvr = \frac{Vr}{B}\frac{dPr}{dt} - Ar\frac{dx}{dt} - Q_l \tag{8}$$

A force balance on the actuator yields:

$$M\frac{d^2x}{dt^2} = PcAc - PrAr - C\frac{dx}{dt} - (Fst + Fstk + Fcol)$$
(9)

At the point of zero actuator velocity, the fluid pressure must overcome the breakout or sticking force, Fstk, before motion is established. At that time a constant force due to Coulomb friction, Fcol, opposes the direction of motion. These non-linear terms have the effect of increasing system damping.

#### 2.2 Servovalve

The servovalve spool actuator is modeled as a first order system with response given as:

$$\tau \frac{dy}{dt} + y = Y_{cmd} \tag{10}$$

with

$$\tau = 1/(2\pi f)$$

Servovalve bandwidth, f, is typically specified by the manufacturer. The oil flow of the servovalve is a function of the square root of differential pressure across the valve and is given as;

$$Qvc = Kv(P_{vsp} - P_{vb})^{1/2} (11)$$

$$Qvr = Kv(P_{va} - P_{vrt})^{1/2} (12)$$

The variable, Kv, is computed from manufacturer ratings of flow for a given differential pressure. For instance, if the servovalve is rated for 40 gpm at a pressure loss of 1000 psi, Kv would be calculated as:

$$Kv = (3.85)(40)/(1000)^{1/2} = 4.87in^4/(sec(lbf)^{1/2})$$
 (13)

A servovalve spool may be manufactured with overlap to insure positive shutoff. An overlapped spool results in a deadband about the neutral position and acts to increase system damping. Deadband is modeled with an ACSL command by setting the dependent variable to zero when the independent variable lies between a specified lower and upper bound. (ref. 2)

#### 2.3 Accumulator

The accumulator is charged with gas which is assumed to behave isentropically. Accumulator pressure is shown to be:

$$P_{accin}(Vgin)^{\gamma} = P_{acc}(Vg)^{\gamma} \tag{14}$$

$$Vg = V_{tot} - V_{lq} \tag{15}$$

$$\frac{dV_{lq}}{dt} = Q_{sp} \tag{16}$$

$$V_{lq} = \int Q_{sp} dt \tag{17}$$

$$P_{acc} = P_{accin}[Vg_{in}/(V_{tot} - \int Q_{sp}dt)]^{\gamma}$$
(18)

#### 2.4 Piping

Pressure losses in the innerconnection piping is modeled by use of pressure loss coefficients. There are four significant sections of piping; the accumulator to servovalve, servovalve to return, and servovalve to rod and cap ends of the actuator cylinder. These equations are defined as:

$$DP_{sp} = K_{sp}(Q_{sp})^2 \tag{19}$$

$$DP_c = K_c(Q_c)^2 (20)$$

$$DP_r = K_r(Q_r)^2 \tag{21}$$

$$DP_{rt} = K_{rt}(Q_{rt})^2 \tag{22}$$

#### 2.5 Transfer Function

The system of equations for the generalized hydraulic actuator system may be linearized by making the following assumptions.

- 1. Coulomb friction and sticking are negligible
- 2. Pressure losses in the piping system are negligible
- 3. Supply pressure remains constant
- 4. Zero servovalve overlap

Incorporating these assumptions, a block diagram of the linearized system is shown in Figure

2. The block diagram may be reduced to yeild the system transfer function:

$$\frac{X}{Y_{cmd}} = \frac{BK_v \sqrt{\frac{P_{sp}}{2}} \left(\frac{Ar}{Vr} + \frac{Ac}{Vc}\right) / M}{S(\tau S + 1) \left[S^2 + \frac{C}{M}S + \frac{B}{M} \left(\frac{A_r^2}{Vr} + \frac{A_c^2}{Vc}\right)\right]}$$
(23)

Which may be put in the form;

$$\frac{X}{Y_{cmd}} = \frac{W_N^2 K v \left(\frac{A_r V_c + A_c V_r}{A_r^2 V_c + A_c^2 V_r}\right) \sqrt{\frac{P_{sp}}{2}}}{S(\tau S + 1) \left[S^2 + 2\xi W_N S + W_N^2\right]}$$
(24)

yeilding system natural frequency and damping.

$$W_N^2 = \frac{B}{M} \left( \frac{A_r^2}{V_r} + \frac{A_c^2}{V_c} \right) \tag{25}$$

$$\xi = C/(2W_N M) \tag{26}$$

For the case  $A_r = A_c$  and  $V_r = V_c$ , such as a double acting linear actuator operating about midstroke the transfer function would reduce to:

$$\frac{X}{Y_{cmd}} = \frac{W_N^2 K v \sqrt{\frac{P_{sp}}{2}} / A}{S(\tau S + 1) \left[ S^2 + 2\xi W_N S + W_N^2 \right]}$$
(27)

$$W_N^2 = \frac{2BA^2}{VM} \tag{28}$$

$$\xi = C/(2W_N M) \tag{29}$$

A rotary motor can also be modeled if the mass of the load is replaced with inertia of the rotary motor.

#### 3.0 Method of Solution

A computer model of the above equations (Appendix 7.0) was implemented using the ACSL code. ACSL was written to facilitate the solution of a system of nth order, nonlinear ordinary differential equations which describe a physical system. Special functions are available in addition to standard Fortran commands to provide flexibility in the problem description. The code sorts the commands into an executable sequence such that a variable is calculated before it is used, allowing the system of equations to be inputed in a logical sequence. Nine integration algorithms are available (default is fourth order Runge-Kutta) in addition to a user supplied routine. The code assigns lower order derivatives a variable name then simultaneously solves the system of first order equations and determines the system states.

A post run command invokes a linear analysis which determines the system poles and damping. The linearized forcing matrix is determined for perturbations about a particular operating point. This capability is particularly useful for designing or developing control strategies.

#### 4.0 Results

A typical use for a servovalve controlled hydraulic actuator system would be that of model injection into the test chamber of a supersonic wind tunnel. Table I shows system parameters for such an application. Dynamic behavior for this model was simulated by solving the nonlinear equations described in section 2 using the ACSL code. Results are presented for three different servovalve command functions.

## 4.1 Case 1, Step Command to Servovalve

Figure 3a-e presents results for a step command to the servovalve. The servovalve was given a step command of 30%, held for 0.5 second than returned to the neutral position. Table I shows the ACSL equation for the step command. Simulated response to this command shows large initial acceleration as the servovalve opens rapidly subjecting the actuator piston to large differential pressure. While the servovalve spool position is held fixed, the actuator piston velocity remains fairly constant, decreasing slightly as accumulator pressure decreases. Upon return of the servovalve spool to the neutral position, the actuator again undergoes large acceleration/deceleration. Oscillation (second order response) of the piston and load occurs due to the compressibility of fluid (fluid bulk modulus). These oscillations are damped by viscous and coulomb friction. The maximum flow for this simulation occurs to the cap end of the actuator, about +15GPM, while the rod end experiences about a -7GPM flow rate (flow in is taken as positive, flow out is negative). The difference in these flow rates is due to the unequal piston area of the rod and cap ends.

# 4.2 Case 2, Steep Ramp Command to Servovalve

Figure 4a-e presents results for a steep ramp command to the servovalve. The servovalve was given a command of 150%/sec for 0.2 second, then held at 30% for 0.2 second and returned to the neutral position at a rate of -150%/sec. Table I shows the ACSL equation for this ramp command. From these results it is seen that the actuator piston and load is subjected to much lower peak acceleration than in Case 1. Maximum velocity and flows are similar due to the same command hold state. Piston oscillation is much less pronounced and of shorter duration than that of Case 1.

#### 4.3 Case 3, Shallow Ramp Command to Servovalve

Figure 5a-e presents results for a shallow ramp command to the servovalve. The servovalve was given a command of 60%/sec for 0.5 second, then held at 30% for 0.1 second, and then returned to the neutral position at a rate of -60%/sec. Table I shows the ACSL equation for this ramp command. Similar to the results of Case 2, peak acceleration of the piston and load is again reduced with the less abrupt servovalve spool movement. Maximum velocity and flows are again about the same due to the same command hold state as the previous simulations. Piston oscillation is almost non-existent for this case.

These simulations indicate that the acceleration of the piston and load are determined by the rate of change of the servovalve command. The magnitude of the servovalve command determine piston and load slew rate.

#### 5.0 Conclusion

A general purpose math model of a servovalve controlled hydraulic actuator system is derived for use with the Advanced Continuous Simulation Language (ACSL). Parameters typical of a hydraulic system were input and the model exercised. The math model is of a general hydraulic actuator system which can be easily changed. This allows for quick assessment of alternate system hardware or control strategy.

## 6.0 References/Acknowledgement

- Burrows, C. R., "Fluid Power Servomechanisms," Van Nostrand Reinhold Co., London, 1972.
- 2. Franklin, G. F., Powell, J. D., Emami-Naeini, A., "Feedback Control of Dynamic Systems," Addison-Wesley Company, Menlo Park, California, 1986.
- Mitchell and Gauthier Associates, "Advanced Continuous Simulation Language," 4th ed., Concord, Mass., 1987.
- 4. Crane Co., "Flow of Fluids", 1981.

Acknowledgement is extended to Mr. J. F. Watson Jr. of NASA Langley Research Center. Mr. Watson provided the author with an ACSL model and schematic of a hydraulic injection system which became the basis of this memorandum. Appendix 7.0 contains the code listing and revisions incorporated into the model.

# Table I Example Simulation Parameters

Actuator			
Cylinder Length Stroke Mass of Piston & Load Cylinder I.D. Rod Diameter Kq <sub>act</sub> Fstk Fcol	17 15 2050 2.0 1.375 0.0 75 50	Inch Inch Lbm Inch Inch Inch Lbf Lbf	
Servo Valve			
Bandwidth Max Flow @ 1000 psid Deadband	25 40 +/-2	Hz GPM %	
Accumulator and Piping			
Volume Initial Gas Charge Supply Pressure Return Pressure Kqsup Kqret Kqcap Kqrod	5 1800 3000 25 0.0125 0.0125 0.0125 0.0125	Gal psia psig psig Lbf sec <sup>2</sup> /in <sup>8</sup> Lbf sec <sup>2</sup> /in <sup>8</sup> Lbf sec <sup>2</sup> /in <sup>8</sup>	

Command to Servo Valve

$$Y_{cind} = 30(step(.2)-step(.7))$$

Case 2. (Figure 4), Steep Ramp Command

$$\mathbf{Y}_{cmd} = 150(\text{ramp}(.2)\text{-ramp}(.4)\text{-ramp}(.6)\text{+ramp}(.8))$$

Case 3. (Figure 5), Shallow Ramp Command

$$\mathbf{Y}_{cmd} = 60(\text{ramp}(.1)\text{-ramp}(.6)\text{-ramp}(.7)\text{+ramp}(1.2))$$

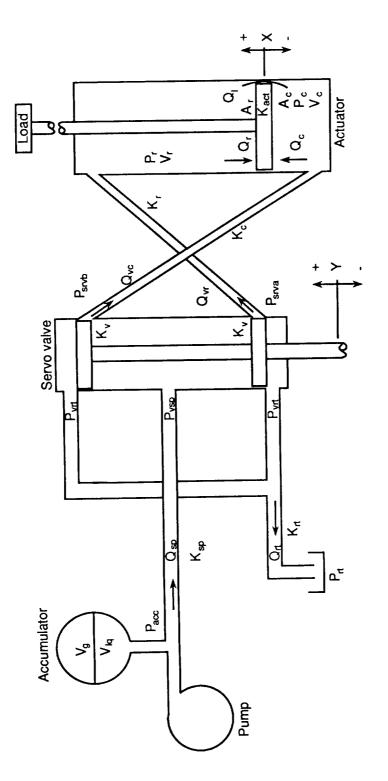
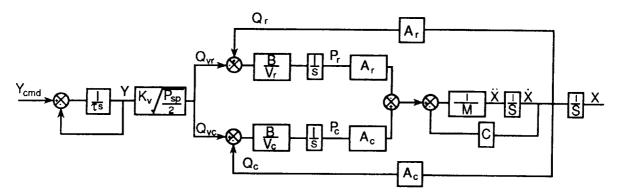


Figure 1. Schematic of Servovalve Controlled Hydraulic Actuator System.



Note:  $P_{\text{sup}} = \text{Supply pressure: Return}$  pressure assumed zero

Transfer function:

$$\frac{X}{Y_{CMD}} = \frac{BK_{V}\sqrt{\frac{P_{SP}}{2}}\left(\frac{A_{r}}{V_{r}}, \frac{A_{c}}{V_{c}}\right)/M}{S(\tau S+1)\left[S^{2} + \frac{C}{M}S + \frac{B}{M}\left(\frac{A_{r}^{2}}{V_{r}}, \frac{A_{c}^{2}}{V_{c}}\right)\right]}$$
For  $A_{r} \stackrel{\sim}{=} A_{c} = A$  and  $V_{r} \stackrel{\sim}{=} V_{c} = V$ 
Transfer function reduces to:

$$\frac{X}{Y_{CMD}} = \frac{2A BK_{v} \sqrt{2P_{sp}} / (VM)}{S(\tau S+1) (S^{2} + \frac{C}{M} S + 2BA^{2} / (VM))}$$

Figure 2. Block Diagram of Linearized Servovalve Controlled Hydraulic Actuator System for Small Perturbations.

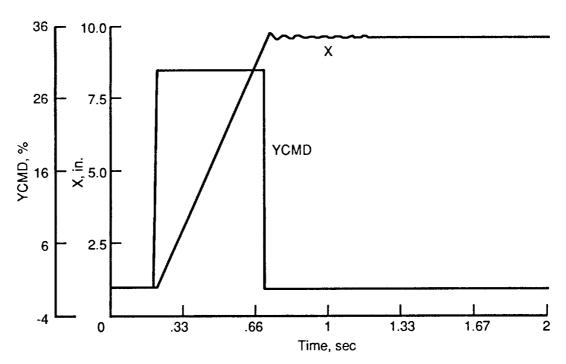


Figure 3a. Servovalve Command (YCMD) and Actuator Position (X) Versus Time for a Step Command to the Servovalve.

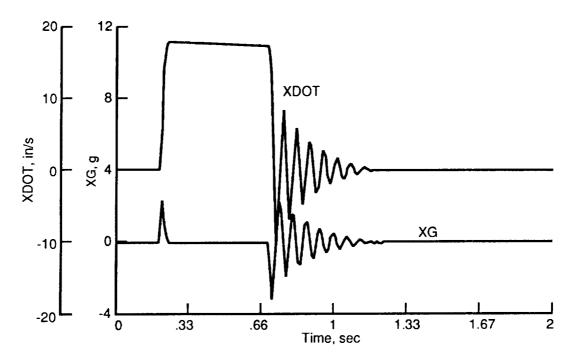


Figure 3b. Actuator Velocity (XDOT) and Acceleration (XG) Versus Time for a Step Command to the Servovalve.

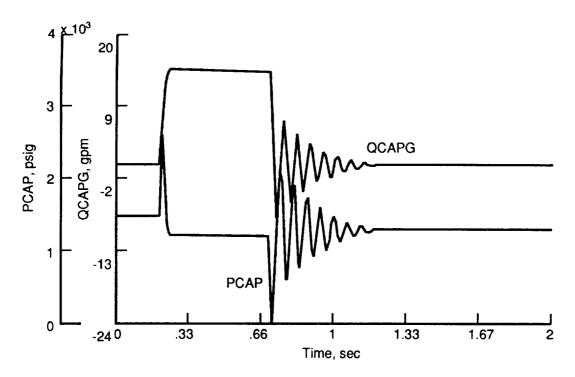


Figure 3c. Cap Side Flow (QCAPG) and Pressure (PCAP) Versus Time for a Step Command to the Servovalve.

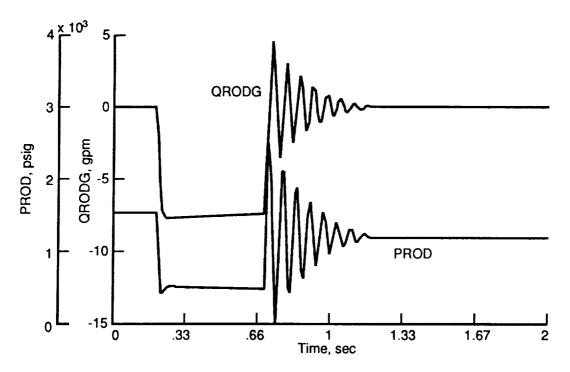


Figure 3d. Rod Side Flow (QRODG) and Pressure (PROD) Versus Time for a Step Command to the Servovalve.

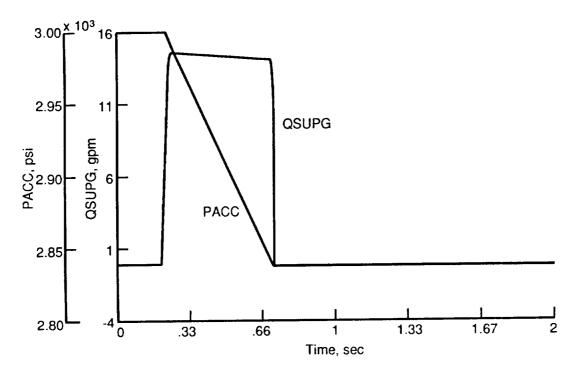


Figure 3e. Supply Pressure (PACC) and Supply Flow (QSUPG) Versus Time for a Step Command to the Servovalve.

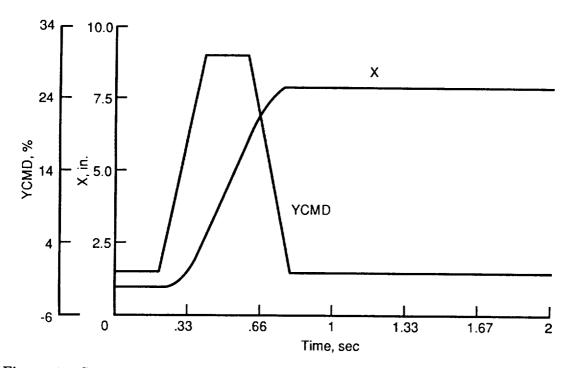


Figure 4a. Servovalve Command (YCMD) and Actuator Position (X) Versus Time for a Steep Ramp Command to the Servovalve.

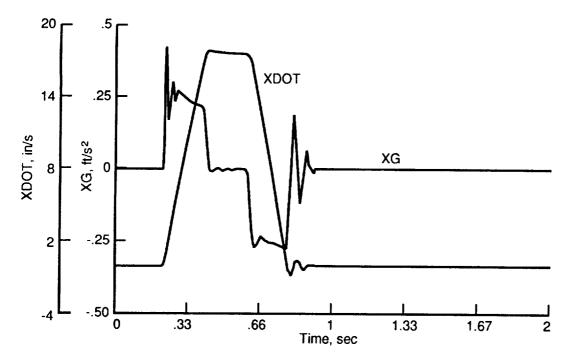


Figure 4b. Actuator Velocity (XDOT) and Acceleration (XG) Versus Time for a Steep Ramp Command to the Servovalve.

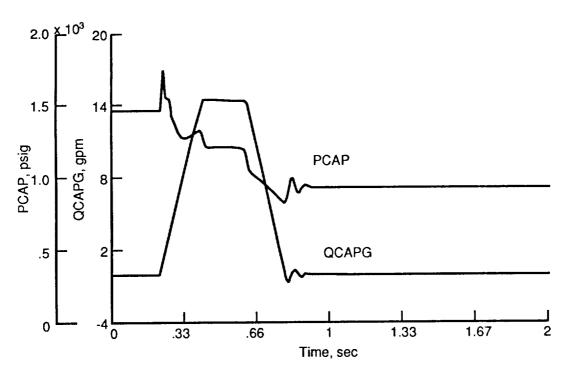


Figure 4c. Cap Side Flow (QCAPG) and Pressure (PCAP) Versus Time for a Steep Ramp Command to the Servovalve.

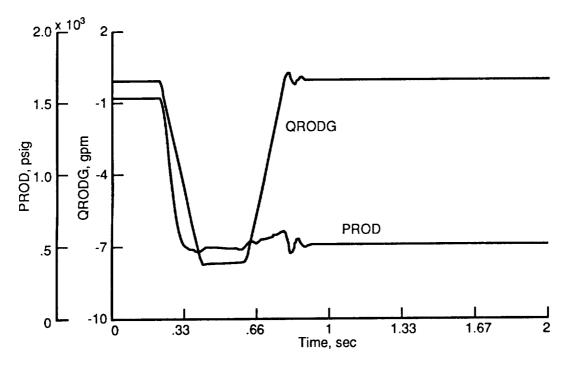


Figure 4d. Rod Side Flow (QROD) and Pressure (PROD) Versus Time for a Steep Ramp Command to the Servovalve.

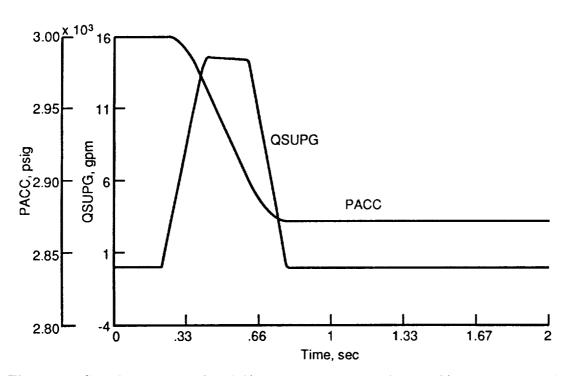


Figure 4e. Supply Pressure (PACC) and Supply Flow (QSUPG) Versus Time for a Steep Ramp Command to the Servovalve.

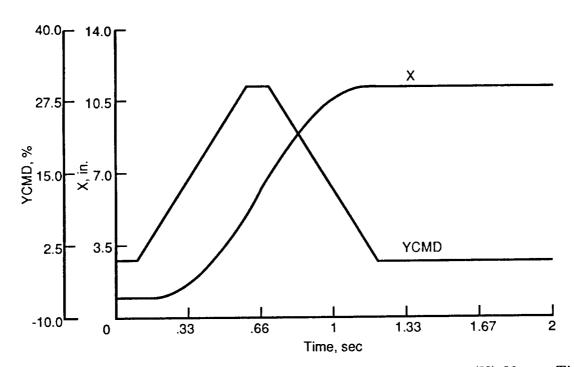


Figure 5a. Servovalve Command (YCMD) and Actuator Position (X) Versus Time for a Shallow Ramp Command to the Servovalve.

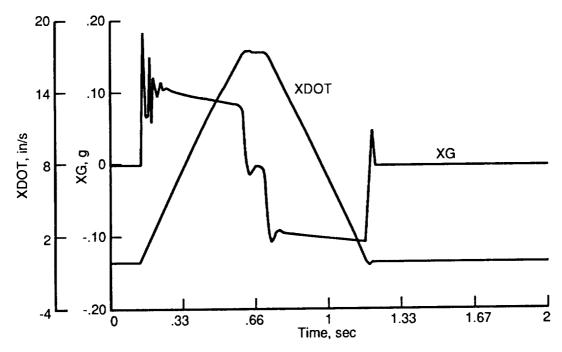


Figure 5b. Actuator Velocity (XDOT) and Acceleration (XG) Versus Time for a Shallow Ramp Command to the Servovalve.

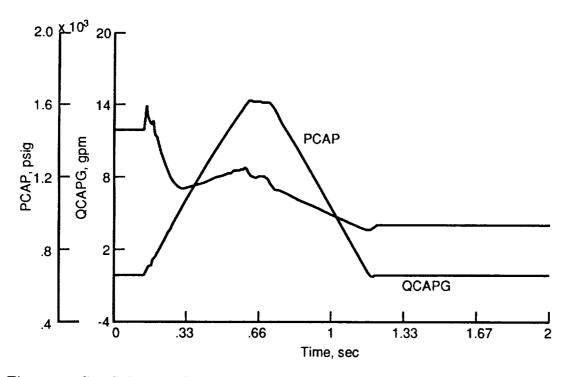


Figure 5c. Cap Side Flow (QCAPG) and Pressure (PCAP) Versus Time for a Shallow Ramp Command to the Servovalve.

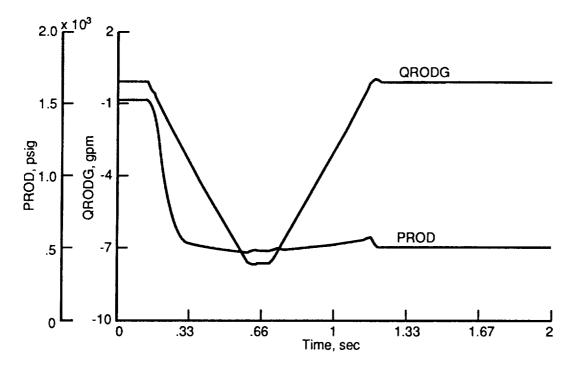


Figure 5d. Rod Side Flow (QRODG) and Pressure (PROD) Versus Time for a Shallow Ramp Command to the Servovalve.

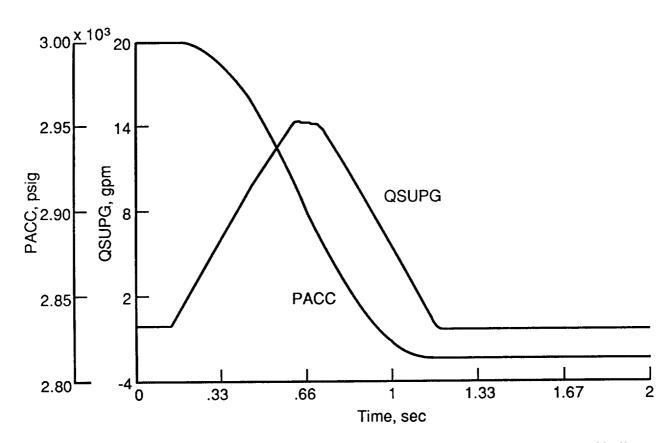


Figure 5e. Supply Pressure (PACC) and Supply Flow (QSUPG) Versus Time for a Shallow Ramp Command to the Servovalve.

## 7.0 Code Listing

#### PROGRAM CYLINDER

SERVOVALVE CONTROLLED HYDRAULIC ACTUATOR SYSTEM ANALYSIS
ORIGINAL VERSION BY J. F. WATSON, FENGD, FACS
REVISED BY

D. A. BABCOCK MAY 1989, FENGD, PSS

THIS UPDATED VERSION OF THE ORIGINAL CODE INCLUDES THE EFFECTS OF ACTUATOR STICKING AT ZERO VELOCITIY; ACTUATOR COULOMB FRICTION; ACTUATOR LEAKAGE FROM THE HIGH TO LOW PRESSURE PORTS; AND SERVOVALVE DEADBAND AS A PERCENTAGE OF OVERLAP: THESE NONLINEAR TERMS INCREASE THE SIMULATED SYSTEM DAMPING

X XDOT XDDOT IX	,	INCH IN/SEC IN/SEC**2 IN
CYSTK CYDIA RODDIA WACTR WLOAD	ACTUATOR STROKE LENGTH ACTUATOR PISTON DIAMETER ACTUATOR ROD DIAMETER ACTUATOR PISTON AND ROD WEIGHT LOAD WEIGHT	IN IN IN LB LB
FSTAT FSTK	VISCOUS DAMPING VISCOUS FRICTION FORCE GRAVITY FORCE OF LOAD AND ACTUATOR ACTUATOR STICKING FORCE ACTUATOR COULOMB FRICTION FORCE	LB LB-SEC/IN LB LB LB LB LB LB
PCAP	ROD SIDE CYLDR PRESSURE CAP SIDE CYLDR PRESSURE LINE PRESSURE LOSS ROD SIDE LINE PRESSURE LOSS CAP SIDE	PSIG PSIG PSID PSID
QROD QCAP VROD VCAP AROD ACAP	ROD SIDE CYLDR OIL FLOW CAP SIDE CYLDR OIL FLOW ROD SIDE CYLDR OIL VOLUME CAP SIDE CYLDR OIL VOLUME ROD SIDE AREA CAP SIDE AREA	IN**3/SEC IN**3/SEC IN**3 IN**3 IN**2 IN**2

```
QL
           ACTUATOR LEAKAGE
                                            IN**3/SEC
      QSUP
            OIL SUPPLY FLOW
                                            IN**3/SEC
      QRET
            OIL RETURN FLOW
                                            IN**3/SEC
      PACC
            ACCUM SUPPLY PRESSURE
                                              PSIG
      POILCG
            INITIAL SYSTEM PRESSURE
                                              PSIG
            RETURN PRESSURE
      PRET
                                              PSIG
     VOLACC
            ACCUMULATOR VOLUME
                                              GAL
     PN2CHG INITIAL NITROGEN CHARGE
                                              PSIA
            SERVOVALVE SPOOL DISP
     YCMD
            COMMAND TO SERVO VALVE
                                               %
     YDOT
            SERVOVALVE SPOOL VELOCITY
                                              %/SEC
     TAU
            SERVOVALVE TIME CONSTANT
                                              SEC
            SERVOVALUE DEAD BAND
     DBAND
                                               %
     SERVEL SERVOVALVE FLOW AT RATED PRESSURE
                                              GPM
     SERVPR SERVOVALVE PRESSURE AT RATED FLOW
                                             PSID
            SERVOVALVE SUPPLY PRESSURE
     PSRVP
                                              PSIG
     PSRVT
            SERVOVALVE RETURN PRESSURE
                                              PSIG
     PSRVA
            SERVOVALVE CONTROL PORT PRESSURE
                                              PSIG
     PSRVB
            SERVOVALVE CONTROL PORT PRESSURE
                                             PSIG
     DPPSUP
            LINE PRESSURE LOSS SUPPLY SIDE
                                              PSID
     DPPRET LINE PRESSURE LOSS RETURN SIDE
                                             PSID
     QVR
            SERVOVALVE ROD SIDE FLOW
                                            IN**3/SEC
     QVC
            SERVOVALVE CAP SIDE FLOW
                                            IN**3/SEC
     KQSER
            SERVOVALVE FLOW COEF
                                        IN**4/(SEC-LB**.5)
     KOSUP
            SUPPLY PIPING LOSS COEF
                                         LB-SEC**2/IN**8
     KQRET
            RETURN PIPING LOSS COEF
                                         LB-SEC**2/IN**8
     KQCAP CAP PIPING LOSS COEF
                                         LB-SEC**2/IN**8
     KQROD ROD PIPING LOSS COEF
                                         LB-SEC**2/IN**8
     KQACT ACTUATOR LEAKAGE COEF
                                         IN**5/(SEC-LB)
     В
            FLUID BULK MODULUS
                                              PSI
     GAMM
            IDEAL GAS SPECIFIC HEAT RATIO
                                              NA
     ATM
            LOCAL ATMOSP PRESSURE
                                              PSIA
----- SERVOVALVE ------
CONSTANT SERVFL = 40.0, SERVPR = 1000.0
CONSTANT DBAND = 0.02, TAU = 0.006
   ----- ACCUMULATOR AND PIPING ------
CONSTANT PN2CHG = 1800.0, VOLACC = 5.0,
                                      GAMM = 1.4
CONSTANT KQSUP = 0.01250, KQRET = 0.01250
CONSTANT KQCAP = 0.01250, KQROD = 0.01250
   CONSTANT POILCG = 3000.0, PRET = 25.0, B = 100000.0
```

```
CONSTANT CYSTK = 17.0, CYDIA = 2.0, RODDIA = 1.375
 CONSTANT WACTR = 500.0, WLOAD = 2000.0, KFV = 50.0
 CONSTANT FSTK = 75.0, FCOL = 50.0, KQACT = 0.01
 CONATANT ALL = 0.5 $" ACTUATOR LOWER INTEG LIMIT "
CONATANT AUL = 16.5 $" ACTUATOR UPPER INTEG LIMIT "
  ----- OTHER -----
 LOGICAL STUCK
                   $ " TRUE IF VELOCITY IS ZERO OTHERWISE FALSE "
 CONSTANT TSTRT =
                 0.0
 CONSTANT TSTP =
                 1.0
 CONSTANT ATM
            =
                 14.7
 CONSTANT PI =
                 3.1415927
CONSTANT INPGAL =
                 231.0 $" IN**3 PER GALLON "
CONSTANT CNURT =
                 3.85
                       $" IN**3/SEC TO GPM *
CONSTANT ACCG =
                 386.4
                       $" GRAVITATIONAL CNST IN/SEC**2 "
CONSTANT LL
             =
                5.0 $" LOWER PRES INTEGRATION LIMIT "
                5000.0 $" UPPER PRES INTEGRATION LIMIT "
CONSTANT UL
             =
INITIAL $" SET INITIAL CONDITIONS "
"----- ACCUMULATOR ------
VLN20 = VOLACC * INPGAL * PN2CHG / ( POILCG + ATM )
KACCM = (POILCG + ATM) * (VLN20 ** GAMM)
   RESET("NOEVAL")
STUCK = XDOT .EQ. 0.0 $" ACTUATOR STICKS WHEN VELOCITY IS ZERO "
FVISC
     = 0.0
FSTAT = WACTR + WLOAD
MTOTL = FSTAT / ACCG
.
ACAP = (PI / 4.0) * (CYDIA ** 2)
     = ACAP - (PI / 4.0) * (RODDIA ** 2)
AROD
IPROD =
        (POILCG * ACAP - FSTAT) / (AROD + ACAP)
IPCAP = POILCG - IPROD
IX
     = 0.0
   -----"
KQSER = CNVRT * SERVFL / (SQRT( SERVPR ))
END $"OF INITIAL"
DYNAMI C
DERIVATIVE
```

```
YCMD = 0.3*(STEP(.2) - STEP(.7))"
                                                   $"STEP COMMAND"
   YCMD = 1.5*(RAMP(.2)-RAMP(.4)-RAMP(.6)+RAMP(.8))
                                                   $"RAMP COMMAND"
  YCMD = 0.6*(RAMP(.1)-RAMP(.6)-RAMP(.7)+RAMP(1.2))" $"RAMP COMMAND"
 CONTROLLER SECTION ****************
              SERVOVALVE CONTROLLER MAY BE ADDED HERE
 SERVOVALUE *****************
                         DYNAMIC SECTION
YDOT
         (YCMD - YV) / TAU
YU
         LIMINT ( YDOT, 0.0, -1.0, +1.0 )
      =
Υ
         DEAD ( -DBAND, DBAND, YV )
                     FLOW SECTION
FLOWO =
         0.0
FLOW1 =
         Y*KQSER*(SQRT(ABS(PSRVP-PSRVB)))*SIGN(1.,PSRVP-PSRVB)
FLOW2 = Y*KQSER*(SQRT(ABS(PSRVB-PSRVT)))*SIGN(1.,PSRVB-PSRVT)
FLOW3 = -Y*KQSER*(SQRT(ABS(PSRVA-PSRVT)))*SIGN(1.,PSRVA-PSRVT)
FLOW4 = -Y*KQSER*(SQRT(ABS(PSRVP-PSRVA)))*SIGN(1.,PSRVP-PSRVA)
QVR
         FCNSW(Y,FLOW4,FLOW0,FLOW3)
OVC
         FCNSW(Y,FLOW2,FLOW0,FLOW1)
"************ ACTUATOR FORCES AND POSITION *************
X
         LIMINT ( XDOT, IX, ALL, AUL )
         INTEG ( XDDOT, 0.0)
XDOT
XDDOT
         ( FTOTL + FCF ) / MTOTL
FTOTL
         FCYL - FVISC - FSTAT
FVISC
       =
         KFV * XDOT
FCYL
         PCAP * ACAP - PROD * AROD
FCF
         RSW ( STUCK, -FTOTL, FASF)
FIA
         RSW ( STUCK, ABS ( FTOTL ) - FSTK, XDOT )
SCHEDULE STICK .XZ. FIA
"************* ROD SIDE FLOW AND PRESSURE ***************
PROD
         LIMINT ( PDROD, IPROD, LL, UL )
QROD
         AROD * XDOT
PDROD
         (QVR + QROD + QL) * (B / VROD)
      =
VROD.
         AROD * ( CYSTK - X )
DPPROD =
         ( QROD ** 2 ) * KQROD * SIGN ( 1.0, QROD )
PSRVA
         PROD - DPPROD
OL
         KQACT * ( PCAP - PROD ) $" ACTUATOR LEAKAGE "
"************** CAP SIDE FLOW AND PRESSURE **************
PCAP
        LIMINT ( PDCAP, IPCAP, LL, UL )
0CAP
         ACAP * XDOT
```

```
PDCAP =
          ( QVC - QCAP - QL ) * ( B / VCAP )
 VCAP
        =
          ACAP * X
 DPPCAP =
           ( QCAP ** 2 ) * KQCAP * SIGN ( 1.0, QCAP )
 PSRVB =
           PCAP + DPPCAP
 "********* ACCUMULATOR / SUPPLY AND RETURN ************
 DELVOL =
          INTEG ( QSUP, 0.0 )
 VOLN2 =
          VLN20 + DELVOL
 PACC
          KACCM / ( VOLN2 ** GAMM ) - ATM
 DPPSUP =
          ( QSUP ** 2 ) * KQSUP * SIGN ( 1.0, QSUP )
PSRVP =
          PACC - DPPSUP
          FCNSW ( Y, -QROD, 0.0, QCAP )
QSUP
DPPRET =
          ( QRET ** 2 ) * KQRET * SIGN ( 1.0, QRET )
PSRVT
          PRET + DPPRET
      =
       = FCNSW (Y, -QCAP, 0.0, QROD)
QRET
END
            "OF DERIVATIVE"
DISCRETE STICK
PROCEDURAL
STUCK
          .NOT. STUCK .AND. ABS( FTOTL ) .LT. FSTK
          RSW (STUCK, 0.0, SIGN( FCOL, -FTOTL))
FASE
XDOT
          0.0 $" SET VEL 0 IF STICKING "
          "OF PROCEDURAL"
END
       $
END
          "OF DISCRETE"
"***************** ENGINEERING VARIABLES ***************
" CONVERT IN**3 TO GALLON PER MINUTE
QCAPG
       =
            QCAP
                 /
                     CNURT
QRODG
        =
            QROD
                     CNURT
QSUPG
        =
            QSUP
                 1
                     CNVRT
ORETG
        =
            QRET /
                     CNURT
" CONVERT IN/SEC**2 TO G
XG
           XDDOT /
                    ACCG
TERMT (T .GE. TSTP)
END
      $
          "OF DYNAMIC"
END
      $
           "OF PROGRAM"
```

National Aeronautics and Scace Agrinistation	Report Docume	ntation Page		
1. Report No.	2. Government Accession	No.	3. Recipient's Catalog	No.
NASA TM-101644				
4. Title and Subtitle			5. Report Date	
			June 1990	
Simulated Dynamic Response of a Servovalve Controlled Hydraulic Actuator			6. Performing Organiz	ration Code
7. Author(s)			8. Performing Organiz	ration Report No.
Dale A. Babcock				
Jare A. Babeock			10. Work Unit No.	
			505-80-11-0	5
9. Performing Organization Name and Addres	SS	•	11. Contract or Grant	
NASA Langley Research Cent	er			
Hampton, VA 23665-5225			13. Type of Report and	d Period Covered
12. Sponsoring Agency Name and Address			13. Type of Report and	a remod covered
National Aeronautics and S	nace Administrat:	ion	Technical M	
Washington, DC 20546-0001	page manifestation as		14. Sponsoring Agence	y Code
			<u>.</u>	
16. Abstract				
A general purpose math mod is derived. The system co a single stage servovalve, interconnecting piping. T Continuous Simulation Lang response characteristics. response characteristics w	nsists of a lined a gas charged hy he state equation uage (ACSL) for d Using this gener	ar actuator windraulic accumuls are integral determining the ralized hydrau	th unequal pisulator, and the ted using the system's dyn	ton areas, e Advanced amic ystem model,
17. Key Words (Suggested by Author(s))		18. Distribution Statem	nent	
Dynamic Simulation		Unclassified - Unlimited		
Servovalve Controlled Actuator		Subject Category 31		
19. Security Classif. (of this report)	20. Security Classif. (of th	is page)	21. No. of pages	22. Price
Unclassified	Unclassified		29	A03

•		