N91-10852

COMPUTATIONS OF THREE-DIMENSIONAL STEADY AND UNSTEADY VISCOUS INCOMPRESSIBLE FLOWS

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The INS3D family of computational fluid dynamics computer codes is presented. These codes are used to as tools in developing and assessing algorithms for solving the incompressible Navier-Stokes equations for steady-state and unsteady flow problems. This work involves applying the codes to real-world problems involving complex three-dimensional geometries. The algorithms utilized include the method of pseudocompressibility and a fractional step method. Several approaches are used with the method of pseudocompressibility including both central and upwind differencing, several types of artificial dissipation schemes, approximate factorization, and an implicit line-relaxation scheme. These codes have been validated using a wide range of problems including flow over a backward-facing step, driven cavity flow, flow through various type of ducts, and steady and unsteady flow over a circular cylinder. Many diverse flow applications have been solved using these codes including parts of the Space Shuttle Main Engine, problems in naval hydrodynamics, low-speed aerodynamics, and biomedical fluid flows. The presentation details several of these including the flow through a Space Shuttle Main Engine inducer, vortex shedding behind a circular cylinder, and flow through an artificial heart.

OUTLINE

⊙ Objective and Approach

Summary of Flow Codes
INS3D Family of codes
CENS3D

Applications and Results

Space Shuttle Main Engine (SSME) componentsArtificial Heart Flow

○ Summary and Future Work

Movie
Circular cylinder vortex shedding
Artificial heart flow

OBJECTIVE AND APPROACH

○ Objective

• To develop CFD capability for simulating steady and unsteady viscous incompressible flows (Incompressible Navier-Stokes)

O Approach

• Develop and assess algorithms, and implement in codes

• Develop / implement physical models for engineering analysis (turbulence, cavitation, porous medium, etc.)

• Apply the codes to real-world problems

SUMMARY OF INS3D

⊙ Governing equations

- Incompressible Navier-Stokes equations in generalized 3-D coordinates for steady-state solutions
 - Pseudocompressibility approach

⊙ Numerical scheme

- Finite difference, central differencing plus artificial dissipation Approximate Factorization Single or multiple zones

Turbulence Models Algebraic models k - ε model

O Applications

• Numerous SSME related simulations

⊙ Status

- Distributed to numerous users across the nation
 Available through COSMIC

EXTENSIONS TO INS3D

⊙ INS3D family of research codes used to study various approaches to solving the INS equations in generalized 3-D coordinates

Pseudocompressibility Approach

 \odot INS3D-UP (Steady-State and Time-accurate calculations)

Upwind flux-difference splitting of uniformly high order
Line-relaxation implicit scheme

Characteristic boundary conditions

INS3D-LU (Steady-State and Rotating reference frame)

Finite-volume method

Spectral radius or flux-difference split based dissipation LU-SGS Implicit Scheme

Non-reflecting boundary conditions

Completely vectorized

EXTENSIONS TO INS3D, continued

Fractional Step Approach

○ INS3D-FS (Time-accurate problems)

• Finite-volume method on a staggered mesh

Accurate treatment of geometry

⇒ Exact discrete mass conservation Two-step fractional step method:

Correct for pressure and velocity (Poisson equation) Solve momentum equations in time (AF scheme)

SUMMARY OF CENS3D CODE

⊙ Governing Equations

• Compressible Euler and Navier-Stokes equations and species transport equations in generalized 3-D coordinates

Numerical Methods

Fully-coupled and implicit thermal-chemical nonequilibrium

finite-rate-chemistry Finite volume / flux-limited TVD, optional high-order flux dif-

ference split upwind scheme • LU-SGS implicit scheme

O Applications

• SSME preburner, main combustor and nozzle

⊙ Status

• Research code

VALIDATION CASES

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• Internal flow: Channel, Backward-facing step, Rectangular duct, Turn-around duct

External flow: Circular cylinder (steady-state), Ogive cylinder

• Juncture flow: Cylinder-plate, Wing-plate, Cavity

○ INS3D-UP

Internal flow: Driven cavity, Backward-facing step, Square duct with a 90° bend
 External flow: Oscillating plate, Circular cylinder (steady and

vortex-shedding flows)

○ INS3D-FS

Internal flow: Driven cavity

External flow: Impulsively started circular cylinder, vortex shedding from a circular cylinder

SUMMARY OF APPLICATIONS

• Space Shuttle Main Engine (SSME) (NASA/MSFC, Rocketdyne) Hot Gas Manifold, Main Injector

Bearing

Impeller/Inducer

Preburner

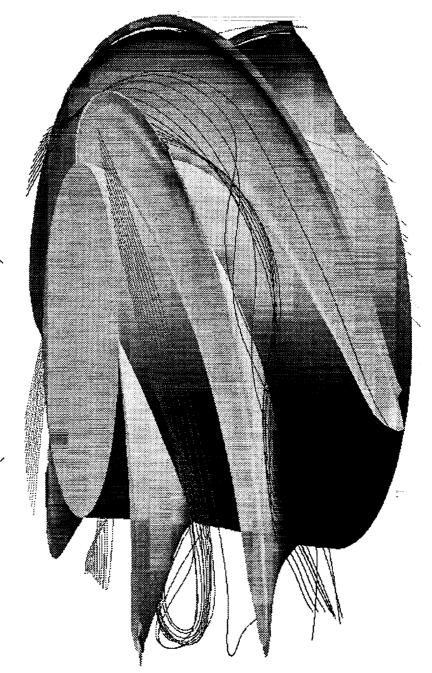
Artificial Heart / Biofluid Mechanics (NASA Tech Utilization, Penn State Univ and Stanford Univ)

Low Speed Aerodynamics High lift device

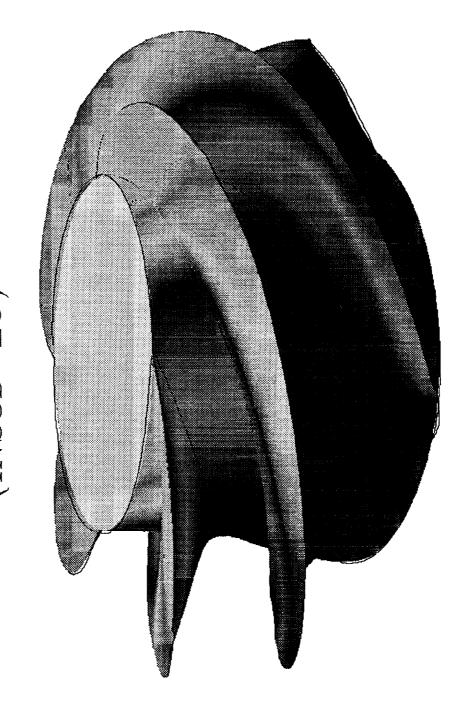
External flow over Automobiles and Trucks

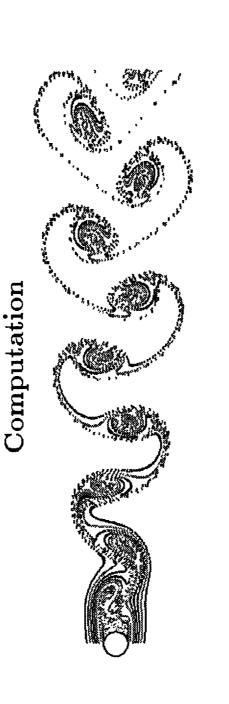
(DARPA, ONR and David Taylor Research Center) Naval Hydrodynamics (Submarine)

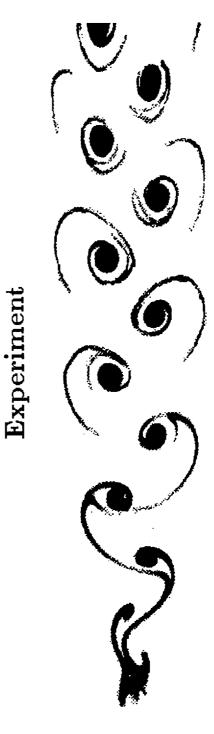
Particle Traces for SSME Inducer (INS3D-LU)



Surface Pressure for SSME Inducer (INS3D-LU)



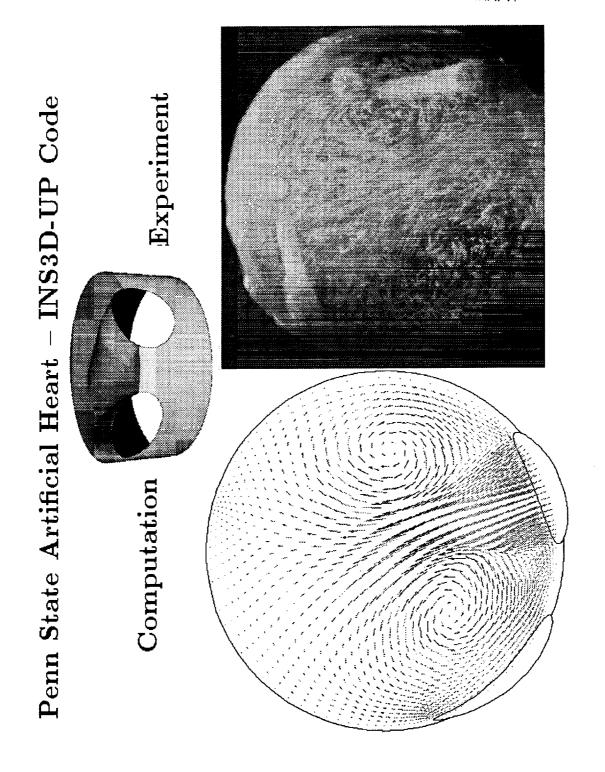




ARTIFICIAL HEART

- Current artificial devices have problems stemming from fluid dynamic phenomena
- High shear stress damages the red blood cells and arterial walls
 - Stagnation and secondary flow regions lead to clotting Desire short residence time in artificial environment
 - - Large pressure losses cause heart to work harder
- © Apply CFD technology to analyzing blood flow through artificial hearts and to suggest improved design
- Develop moving boundary capability
- Apply time accurate flow solvers to Penn State Artificial Heart
 - Develop simple non-Newtonian fluid model

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CONCLUDING REMARKS

- ⊙ Incompressible and low speed flow simulation codes have been developed (INS3D-xx, CENS3D).
- Results of computer simulations have made significant impact on analysis and redesign of the SSME power head.
- ⊙ These codes are being extended to analyze other important real world problems.
- Future work includes further enhancement of these codes and improvement in physical modeling. 0