

N91-28121**EFFECT OF GRINDING ON THE FATIGUE LIFE OF TITANIUM ALLOY
(5 Al - 2.5 Sn) UNDER DRY AND WET CONDITIONS***55B-26
266 39
P-15

Partha Rangaswamy, Hendra Terutung, and Shaik Jeelani
Materials Research Laboratory
School of Engineering and Architecture
Tuskegee University, AL

ABSTRACT

The principal factors in the performance of aerospace materials are strength-to-weight ratio, fatigue life, fracture toughness, survivability and, of course, reliability. Machining processes and, in particular, grinding under adverse conditions have been found to cause damage to surface integrity and affect the residual stress distribution in the surface and subsurface region. These effects have a direct bearing on the fatigue life.

In this investigation the effects of grinding conditions on the fatigue life of Titanium 5 Al-2.5Sn were studied. This alloy is used in ground form in the manufacturing of some critical components in the space shuttle's main engine. It is essential that materials for such applications be properly characterized for use in severe service conditions.

Flat sub-size specimens 0.1 inch thick were ground on a surface grinding machine equipped with a variable speed motor at speeds of 2000-6000 rpm using SiC wheels of grit sizes 60 and 120. The grinding parameters used in this investigation were chosen from a separate study. The ground specimens were then fatigued at a selected stress and the resulting lives were compared with that of the virgin material. The surfaces of the specimens were examined under a scanning electron microscope, and the roughness and hardness were measured using a standard profilometer and microhardness tester, respectively.

The fatigue life of the ground specimens was found to decrease with the increase in speed for both dry and wet conditions. For both grit sizes, the fatigue life was lower than that of the virgin material for the dry condition. The fatigue life of specimens ground under wet conditions showed a significant increase at the wheel speed of 2000 rpm for both the grit sizes and thereafter decreased with increase in speed to below that of the virgin material. The results of the investigation are explained using profilometry, microhardness measurements and scanning electron microscopic examination.

* Work supported by NASA Grant NAG8-068

EFFECT OF GRINDING ON THE FATIGUE LIFE OF TITANIUM ALLOY (5 Al - 2.5 Sn) UNDER DRY AND WET CONDITIONS

Introduction

Titanium alloys are extensively used in aerospace applications because they meet the design and service requirements. It is observed that during manufacturing the consistent preservation of their inherent strength is difficult to achieve. Catastrophic failures have resulted from damage to the surface integrity by machining during manufacturing. Therefore, manufacturing procedures are required to be specified to avoid damage to the surface and the sub-surface, and, if possible, to specify processing which will enhance part performance. This makes it imperative for the design engineer to possess complete information about the surface characteristics of a component and the mechanical properties of the bulk material before the component is designed so that manufacturing procedures could be suggested to withstand service conditions such as fatigue, creep and stress corrosion cracking [1-3]. Previous investigations have shown that in the machining of metals a damaged surfaced region is produced that is different from the bulk of the material [4-7]. Changes in material surface properties can, in turn, influence mechanical properties of which fatigue life and stress corrosion resistance are usually of the most concern [8-11].

Of the various material removal processes studied to date, grinding causes the most impact on material properties [12,13]. Grinding is an abrasive machining process (more precisely termed micro-machining process) and it differs from machining in that grinding speeds are much faster, depth of cut smaller and there is very little quantitative information on the geometry of individual grits at the periphery of grinding wheels. Studies on surface alterations due to grinding have revealed that the extent of deformed layer depends on the size of the grit as well as its rake angle, grinding speed and depth of cut [14]. Studies carried out on grinding of titanium and titanium alloys (Ti-6Al-4V) have also revealed that a large scale redeposition of metal on the surface accounts for poor surface finish at high speeds using silicon carbide (SiC) and aluminum oxide (Al_2O_3) grinding wheels [15]. Also, the high temperatures developed at the wheel-work interface due to the vast range of process parameters involving rapid rates of material removal have led to lowering the level of surface integrity and producing extremely complex pattern of residual stress distribution [8,12]. These factors have a detrimental effect on the fatigue strength and the fatigue life of the components.

The objective of this research was to investigate the effects of grinding on the fatigue life of Ti-5Al-2.5Sn at room temperature under dry and wet conditions. There are no data available at the test range considered in this work to show what effect, if any, grinding has on the fatigue life of a Ti-5Al-2.5Sn alloy at various grinding speeds and for various grit sizes.

Experimental Work:

A) Specimen Preparation

The Ti-5Al-2.5Sn alloy used in this investigation is an alpha type alloy and is the most widely used commercial alloy in this group. It exhibits excellent fracture toughness and corrosive resistance and is widely used for structural and cryogenics applications. Mechanical properties and chemical composition of the material are given in Tables 1 and 2, respectively.

The sub-sized specimen and the loading grips used in this investigation to hold it were selected from the work done earlier by Jeelani et al [17]. The specimen was designed with 0.1 inch gauge section width and 1.0 inch gauge section length. The specimen thickness was 0.1 inch. These specimens were made as per the specification shown in Figure 1, on a numerically controlled machining center. These specimens were made out of a 2-inch long flat plate, 0.75 inch wide and 0.125 inch in thickness.

After the specimens were fabricated, they were annealed by heating and holding at 1400° for 60 minutes and at 1600° for 10 minutes and then cooling in air. Stress relieving was performed by soaking the specimens for 1 hour at 1000°F. However, it should be noted that Ti-5Al-2.5Sn cannot be hardened by heat treatment. It is used only in annealed condition [18].

The oxidized layer and the toolmarks and other surface irregularities were removed by wet grinding/polishing using silicon carbide emery papers of sizes ranging from 60 to 4000 grits per square inch.

B) Grinding Tests

The grinding of all the specimens was carried on a precision surface grinding machine, (BOYAR-SCHULTZ 1A618 Hydraulic Surface Grinder), equipped with a 2 hp variable speed motor (0-6000 rpm) powered by Volkmann Drives 2 HP Adjustable drive unit which converts the fixed input AC frequency to the motor to a variable frequency range, thus changing the usually fixed motor speed to variable speed.

Grinding was carried out with silicon carbide (39C60H8VK and 39C120I8VK) wheels. The wheels were of 6 inch diameter and 1/2 inch width received in dressed condition (coarse dressing). Grinding wheel speeds of 2000 - 4000 - 6000 rpm were used and a table speed of 40 rpm. All grinding tests were made in a single pass. Table 3 shows a summary of the grinding conditions used in this investigation. These conditions are based on the recommendations of the METCUT [12].

C) Fatigue Testing

For testing the virgin and the ground specimens a direct tension-compression fatigue testing machine (Fatigue Dynamics Model DS-6000 HLM) which is equipped with a hydraulic load maintainer was used. It adjusts the preload continuously to preset value without affecting the cyclical load. The cyclical load can be adjusted manually up to 6000 pound force. The test frequency of the machine ranges from 600 to 2200 cycles/minute. A load cell is provided to read tensile or compressive load on the specimen, directly. The load cell is connected to the strain indicator which is calibrated to read the load on the specimen. A stress ratio of $R = 0$ was used in this investigation. Throughout the test, the cycle speed was maintained constant at 200 cycles per minute. Tests were performed at room temperature.

D) Surface Examination

Small pieces approximately 0.25 X 0.25 inch were cut from all virgin and ground specimens. The pieces were cleaned in an aqueous methanol solution, and were then air dried. The ground surfaces were examined by a scanning electron microscope over a wide range of magnification.

A Bendix profilometer group 7L equipped with a Sheffield Profilometer, Model QED-6, Digital amplifier, Sheffield Profilometer type VEG Gated Pilator and type LK tracer with Ft skidmount was used to measure the roughness of the surfaces produced. A tracer stroke of 0.5 inch was used with a selected cut-off of 0.0125 inch. The roughness measurements were made on the specimens in the direction parallel to the work-tool motion. Surface roughness measurements of the parent metal specimens as well as those of the ground specimens were taken.

Microhardness measurements using Buhler Micromet II Digital Microhardness tester were made on the ground surfaces produced at all wheel peripheral speeds. At least five measurements were made on each specimen and the average taken for comparison.

Results and Discussion

Figure 2 shows the S-N diagram generated by using 30 polished specimens of Ti-5Al-2.5Sn alloy. Approximately five specimens were tested at each stress level and the fatigue life data were obtained at five stress levels. A software called Grapher especially used for producing two-dimensional graphs was used to produce the S-N curve. Point REF in the diagram, which corresponds to a stress of 100,000 psi and the fatigue life of 125,000 cycles was used as a reference for comparison of the lives of the specimens ground under various conditions used in this investigation.

Figure 3 shows the effect of the type and speed of the grinding wheel and the presence or absence of the cutting fluid on the fatigue life of the specimens. Each data point on the graph represents an average of five tests. For the specimens ground with wheels of both grit sizes (60 and 120) the fatigue life decreases with an increase in the speed. This trend is consistent with the data published in previous studies [8,19,20]. The fatigue life of the specimens ground under dry conditions was lower than that of the virgin specimens. Dry grinding is also termed abusive grinding, which has been shown to cause severe damage to the surface integrity and affect the residual distribution at the surface, lowering the fatigue strength, therefore resulting in lowered fatigue life. Published data have indicated that dry grinding generally produces tensile stress near to the surface under conditions far from low stress condition [7,21,22].

Figure 4 shows the variation in the surface roughness due to a change in the type and speed of the grinding wheel and the presence or absence of the cutting fluid. It can be seen that for all the grinding conditions used in this investigation the surface roughness is higher than that of the virgin material, and increases as the grinding speed is increased. The presence of cutting fluid in the grinding region has decreased the surface roughness, but the change does not appear to be significant. The graph shows that the surface roughness changes due to a change in the grit size of the grinding wheel, but the change is not significant enough to establish a trend.

Figure 5 shows the variation in the microhardness due to a change in the type and speed of the grinding wheel and the presence or absence of the cutting fluid. For all the grinding conditions used in this investigation it can be seen that the microhardness values show a decrease in hardness with an

increase in grinding speed. The presence of cutting fluid slightly increases the hardness values of the specimens ground under wet conditions as compared with the specimens ground dry. At the low grinding speed of 2000 fpm the hardness of the specimens ground under wet conditions is higher than that of specimens ground dry. This trend is consistent up to 4000 fpm but thereafter it is observed that with an increase in the grinding speed the specimens ground with the wheel of grit size 120 for both the dry and wet conditions show slightly higher hardness values than those of the specimens ground with the wheel of 60 grit size. Here again, the graph shows that the microhardness changes due to a change in the grit size of the grinding wheel, but again, the change is not significant enough to establish a trend.

The scanning electron micrographs of the surfaces produced under both dry and wet conditions showed that the surfaces consisted of microcracks and tears showing plowing of the metal by the abrasive action of the grit, evidence of large scale plastic deformation, redeposition and cavities. The texture showed grooves of non-uniform width and spacing for all the grinding speeds [23].

Figures 6a and 6b show scanning electron photomicrographs of the surfaces produced when ground under dry conditions with the wheel of 60 grit size at the lowest and highest grinding speed in the range tested. These photomicrographs reveal a drastic change in texture ranging from long straight grooves parallel to the wheel-work direction to short discontinuous grooves which vary in width and depth with an increase in the grinding speed. This also explains the high roughness values obtained with increase in speed. It was also observed that at a speed of 6000 rpm the ground surfaces revealed cracks and microcracks for both the grit sizes. The cracking occurs because of the generation of heat during the transfer of metal (redeposition) back onto the ground surface at high speeds and subsequent rapid cooling. These cracks and microcracks could be one of the reasons for the decrease in fatigue life of the specimens ground under dry conditions at higher speeds. The effects of rubbing, plowing and cutting, characteristic of the grinding process, were present for all three speeds [12].

Figures 7a and 7b show scanning electron photomicrographs of the surfaces produced when ground under wet conditions with the wheel of 60 grit size at the lowest and highest grinding speed in the range tested. The texture was well defined at the low grinding speed of 2000 fpm indicating that considerably lesser amounts of redeposition on the surface had taken place. Also, the micrographs revealed that the extent of severity of damage caused to the surface was considerably less than the surface ground under dry conditions. But at the speed of 6000 rpm, the surface did reveal cracks transverse to the direction of grinding similar to that of the surfaces ground under dry conditions which had resulted in a considerable decrease in fatigue life. This shows that the presence of cutting fluid, which appreciably reduced the wheel-workpiece temperature and also aided in removing the grinding chips, thereby minimizing redeposition, had virtually no effect in preventing damage to the surface in the form of cracks as a result of increase in process intensity caused by an increase in grinding speed.

Studies have shown that grinding carried out at low speeds in the presence of a lubricant results in producing residual stresses which are compressive in nature in the surface and sub-surface region [12]. For the titanium alloy used in this investigation, there are no data reported yet concerning the nature of the residual stresses in the surface and subsurface due to grinding.

The surface roughness measurements also show that the roughness increases with an increase in the grinding speed, an observation which is supported by the SEM photomicrographs as shown in Figures 6 and 7. It is to be noted as mentioned earlier that this particular alloy cannot be hardened and therefore the high temperatures reached due to heat generated at the work-wheel interface anneal the workpiece, thereby softening the workpiece at the surface and the sub-surface region. This is one of the reasons for the microhardness measurement showing a decrease with an increase in the grinding

speed. Also, visual examination of the surface of the specimens ground under dry condition at high speeds showed evidence of burnt marks.

Conclusion

From the results of the investigation of the effect of grinding conditions on the fatigue life of Titanium 5AL-2.5Sn, the following conclusions may be drawn:

1. The results of the study indicated that fatigue life of the specimens decreased considerably with an increase in the speed. The fatigue life of the specimens ground under dry conditions was lower than that of the virgin specimens for the speed range tested. The fatigue life of the specimens ground with cutting fluid was higher than that of the virgin specimens in the range of 2000 - 4000 rpm.
2. The results of surface roughness measurements over the entire speed range of the ground specimens showed an increase in roughness with an increase in the speed. The roughness values of the specimens ground under dry and wet conditions were higher than those of the virgin specimens. The roughness of the specimens ground under dry conditions was higher than that of the specimens ground under wet conditions at all speeds and for both grit sizes.
3. The microhardness measurements for both the grit sizes and for both the dry and wet conditions used in this investigation showed a decreasing trend in hardness for all the speeds.
4. The grit sizes of the grinding wheel selected for this study showed no significant effect on the fatigue life.

References

1. Ramakrishnan, K. "Surface and subsurface damage in machining of 18% nickel maraging steel, 6242-titanium alloy, and red brass". M.S. Thesis, Tuskegee Institute, 1979.
2. Koster, W.P., Field, M., Fritz, L.J., Gatto, L.R. and Kahles, J.F. "Surface integrity of machined structural components", USAF Technical Report AFML-TR-70-11, Metcut Research Associates Inc., Cincinnati, OH, 1970.
3. Koster, W.P. and Field, M. "Effect of Machining variables on the surface and structural integrity of titanium", Proceedings of North American Metalworking Research Conference, McMaster University, Hamilton, Ontario, Canada, May 14-15, 1973, (Hamilton: McMaster University, 1973), Vol.2-Metal Cutting and Electrical Machining, pp.67-87.
4. Bailey, J.A. and Jeelani, S. "Surface integrity in the machining of annealed 18% nickel maraging steel", SME Trans (1974) pp 174-185.

5. Bailey, J.A. and Jeelani, S. "Determination of subsurface plastic strain in machining using an embossed grid", *Wear* 36 No 2 (1976) pp 199-206.
6. Bailey, J.A. and Jeelani, S. "Surface integrity in the machining of quenched and tempered AISI-4340 steel", *ASME, J Engng Ind* 98 No 3 (1976) p 999.
7. Bailey, J.A. and Jeelani, S. "State of subsurface region in machining solution treated and aged 18% nickel maraging steel" *Wear* 72 (1981) p 237.
8. Koster, W.P. "Surface integrity of machined materials", USAF Technical Report AFML-TR-74-60, Metcut Research Associates Inc., Cincinnati, OH, 1974.
9. Bellows, G. and Kohls, J.B. "Surface integrity of nontraditional material removal processes", Technical Paper MRR76-12, Society of Manufacturing Engineers, Dearbon, MI, 1976.
10. Jeelani, S. and Musial, M. "Effect of cutting speed and tool rake angle on the fatigue life of 2024-T351 aluminium alloy", In *J Fatigue* 6 No 3 (1984) pp 169-172.
11. Jeelani, S. and Collins, M.R. "Effect of electric discharge machining on the fatigue life of Inconel 718", In *J Fatigue* 10 no 2 (1988) pp 121-125.
12. Bellows, G. "Low Stress Grinding", Machinability Data Center Publication No. MDC 83-103, 1983.
13. Cammett, J.T., III. "Fatigue crack initiation in surface ground Ti-6Al-4V", In *Proceedings of the International Conference on Production Engineering, Tokyo 1974 (Part II)*, pp 132-136. Tokyo, Japan: Japan Society of Precision Engineering, 1974.
14. Turley, D.M. and Doyle, E.D. "Factors affecting workpiece deformation during grinding", *Materials Science and Engineering*, 21 (1975) pp 261-271.
15. Turley, D.M. "Factors affecting surface finish when grinding titanium and a titanium alloy (Ti-6Al-4V)", *Wear.*, 104, 1985 pp 323-335.
16. Jeelani, S., Natarajan, R. and Reddy, G.R. "A subsized fatigue specimen", *Int J Fatigue* 8 No 3 (1986) pp 159-164.
17. D.O.D. *Aerospace Structural Metals Handbook*, Vol. 4, Code 3706, pp 1-13.
18. Kahles, J.F., Field, M., Eylon, D. and Froes, F.H. "Machining of Titanium alloys", *Journal of Metals*, 1985 pp 27-35.
19. Imamn M.A. and Gilmore, D.M. "Fatigue and microstructural properties of quenched Ti-6Al-4V", *Met. Trans. (A)* 14A, 1983 pp233.
20. Leverant, G.R., Langer, B.S., Yuen, A. and Hopkins, S.W. "Surface residual stresses, surface topography and the fatigue behavior of Ti-6Al-4V", *Met. Trans. (A)* 10A 1979 pp 251-257.

21. Takemoto, T., Jimg, K.L., Tsakalagos, T., Weissmann, S. and Dramer, I.R. "The importance of surface layer on fatigue behavior on a Ti-6Al-4V alloy." Met. Trans. (A) 14A 1983 pp 127-132.
22. Rangaswamy, Partha., "Effect of grinding on the fatigue life of Ti-5Al-2.5Sn under unlubricated conditions", M.S. Thesis, Tuskegee Institute, 1989.
23. Terutung, Hendra., "Effect of grinding on the fatigue life of Ti-5Al-2.5Sn under lubricated conditions", M.S. Thesis, Tuskegee Institute, 1989.

Acknowledgments

The authors gratefully acknowledge the support of NASA- Marshall Space Flight Center through Grant NAG8-068 for this study.

TABLE 1. MECHANICAL AND PHYSICAL PROPERTIES OF TITANIUM 5AL 2.5 SN

Ultimate Tensile Strength	120 Ksi
Yield Strength (0.2 % offset)	115 Ksi
Hardness (Rockwell C)	36 Rc
Elongation in 2 in.	10 %
Reduction of Area	25 %
Impact, Charpy V	19 Ft.lb
Rupture Stress	108 Ksi
Modulus of Elasticity, tension	16 Mpsi
Modulus of Elasticity, torsion	6.5 Mpsi
Density	0.162 lb/cu in.
Melting Range	2822-3000 °F
Sp. Elec. Resis.	157 10 ⁻⁶ Ohm-cm
Specific Heat	0.125 (Btu/lb°F)
Mean Coeff. of Ther. Exp.,/ °F	5.2 X 10 ⁻⁶

TABLE 2. CHEMICAL COMPOSITION

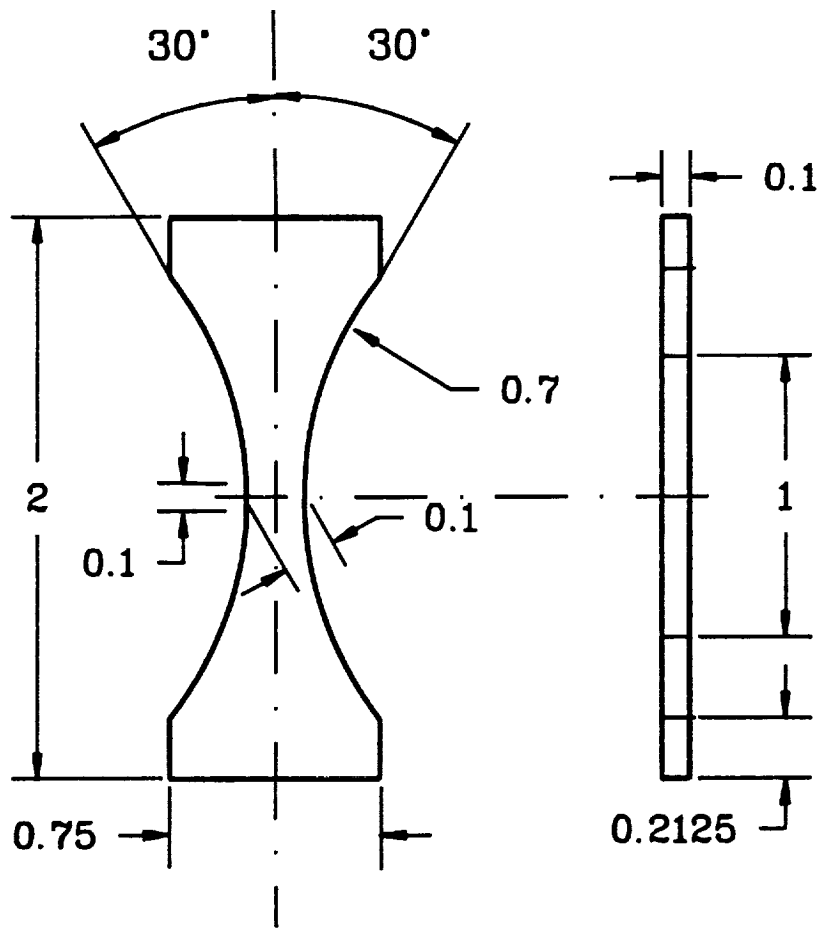
Element	Percent (weight)
Al	4.0 - 6.0
C	0.15
H	0.003 - 0.020
Fe	0.5
Mn	0.3
N	0.07
O	0.2
Sn	2.0 - 3.0
Ti	Balance

TABLE 3. SUMMARY OF GRINDING CONDITIONS

Wheel Type	39C60H8VK, 39C120I8VK
Wheel Speed(fpm)	2000, 4000, 6000
Table Speed(fpm)	50
Down Feed per pass(inch/pass)	0.0005 (16 passes)* 0.0004 (2 passes) 0.0002 (6 passes)**
Cross Feed(inch/pass)	0.05
Wheel Classification	Soft grade (H , I) Open Structure (8) Grain Size (60, 12)

*All Grinding done in Single Pass

**Wheel Dressed before Final Pass (Coarse dressing done to maintain sharpness)



all dimensions in inches

Figure 1. Fatigue specimen.

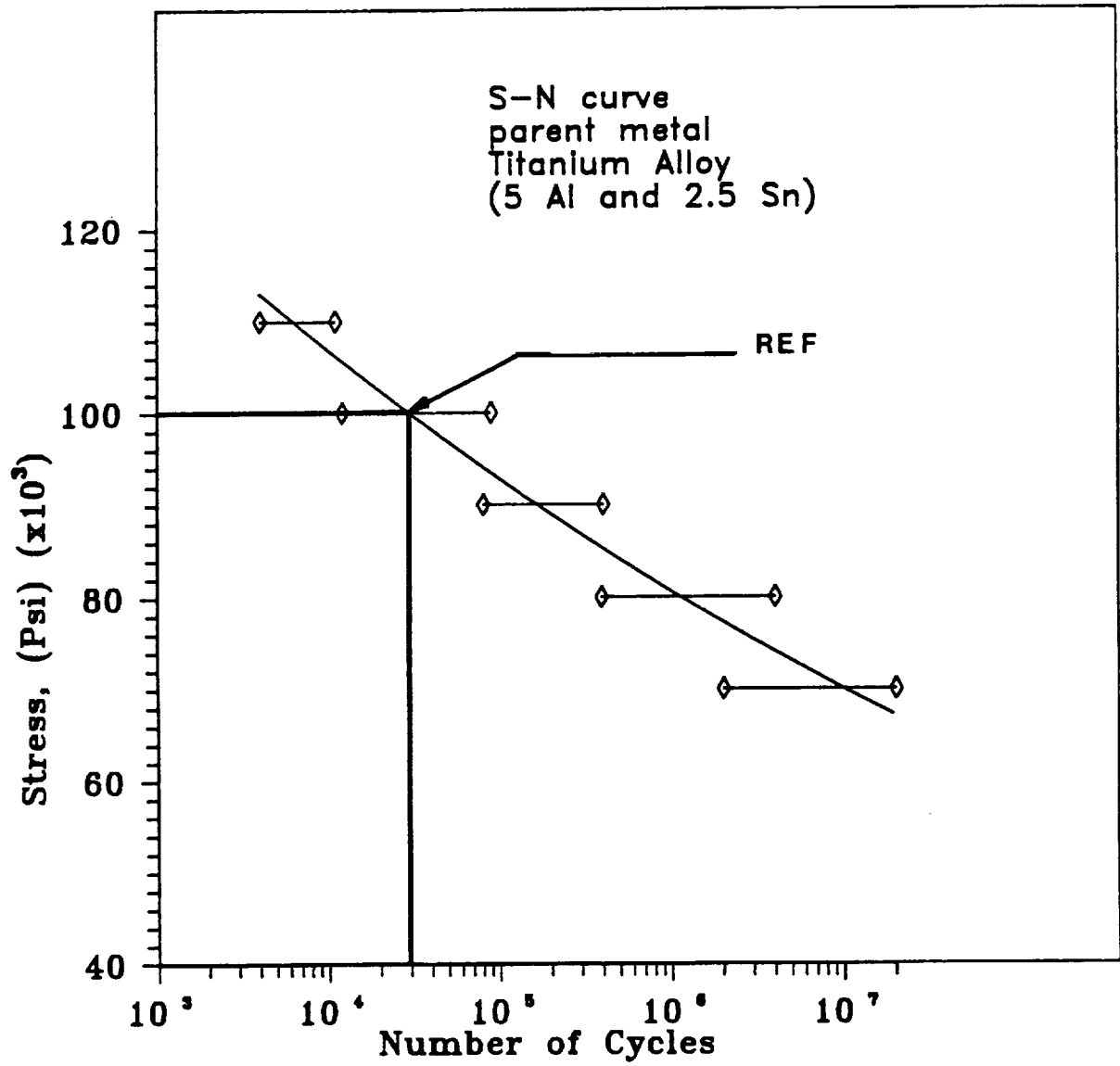


Figure 2. S/N Diagram for Titanium alloy virgin material.

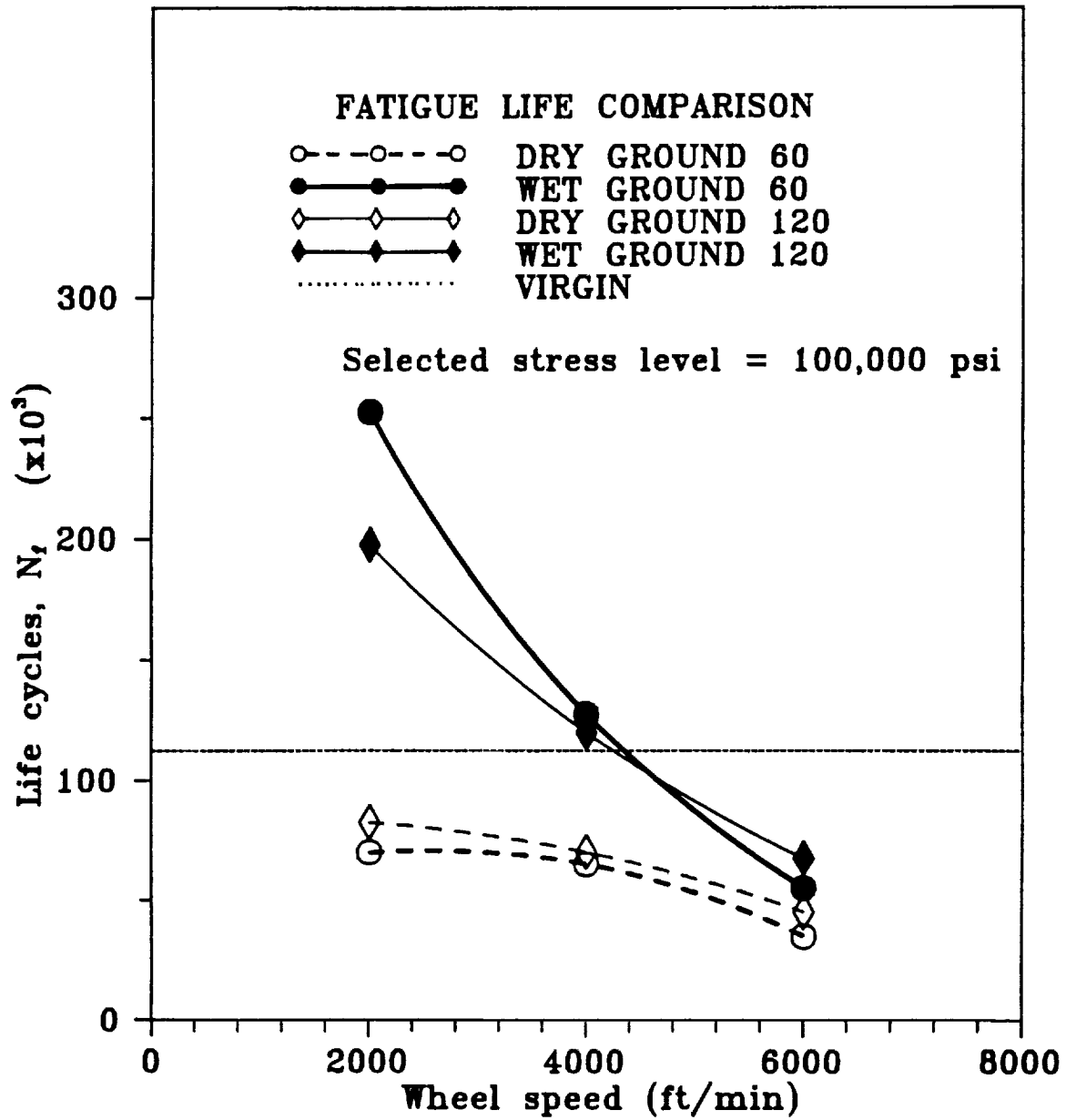


Figure 3. Fatigue life vs wheel speed: virgin material and ground specimens.

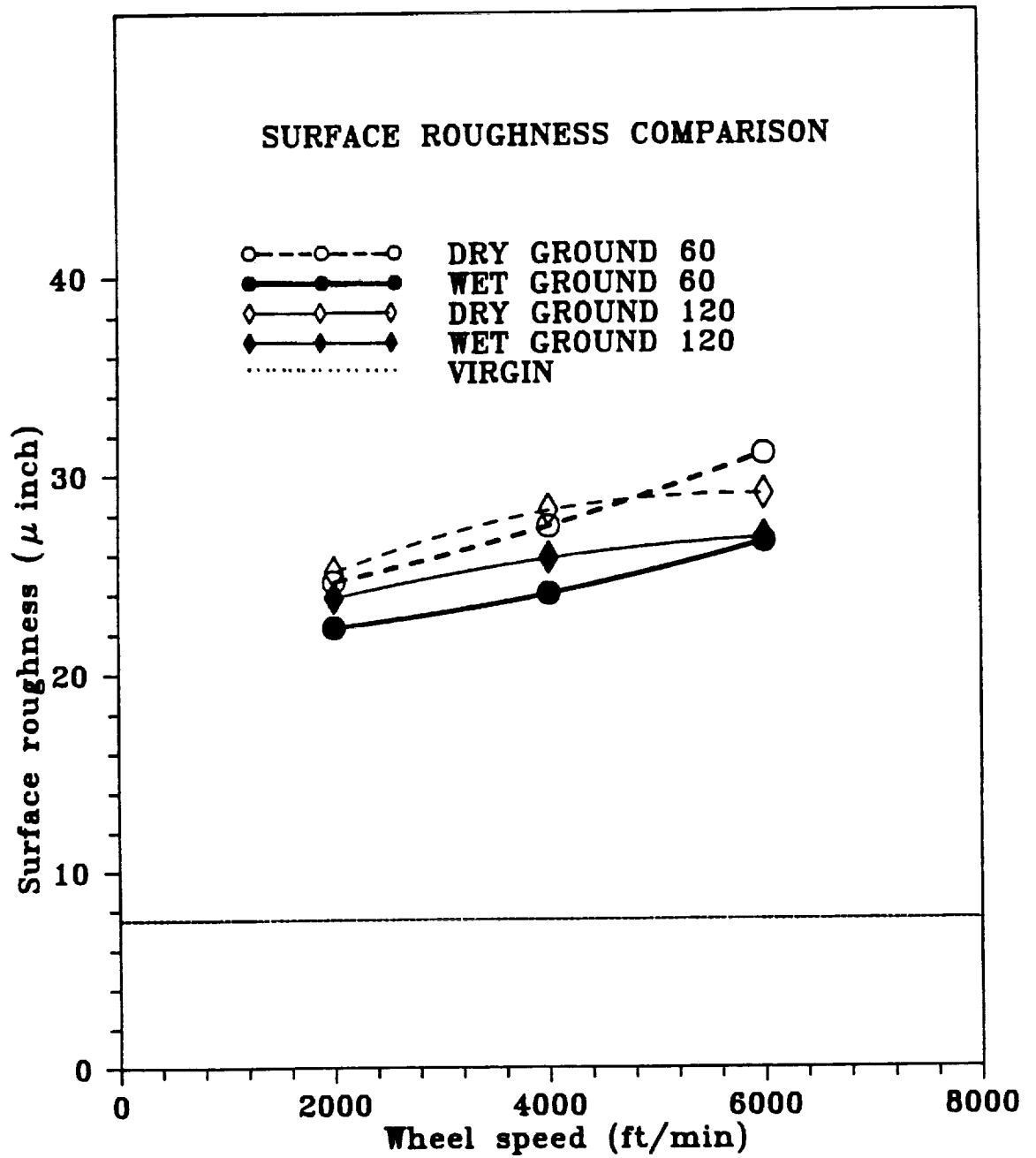


Figure 4. Surface roughness vs wheel speed: virgin material and ground specimens.

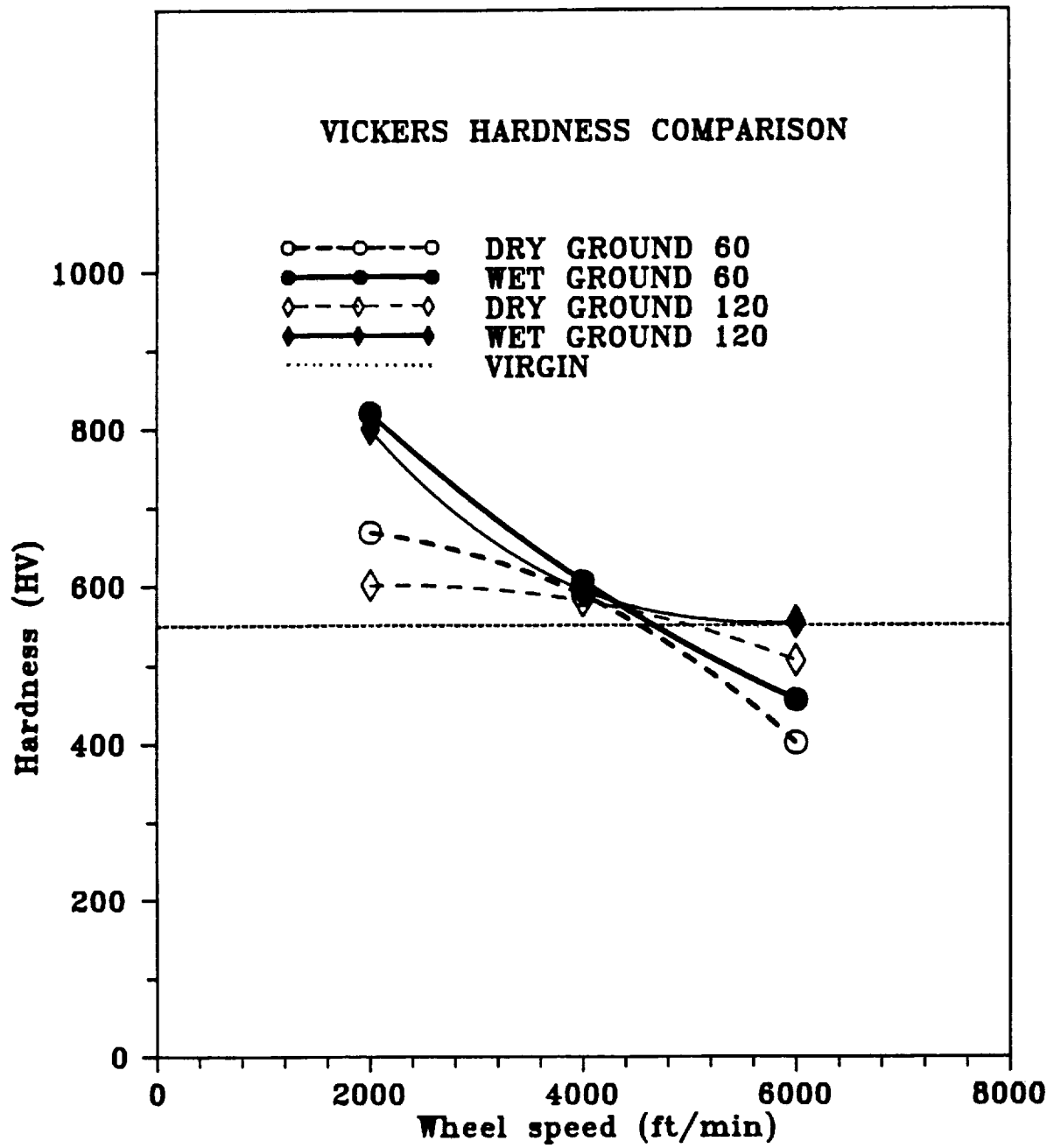


Figure 5. Vickers hardness vs wheel speed: virgin material and ground specimens.

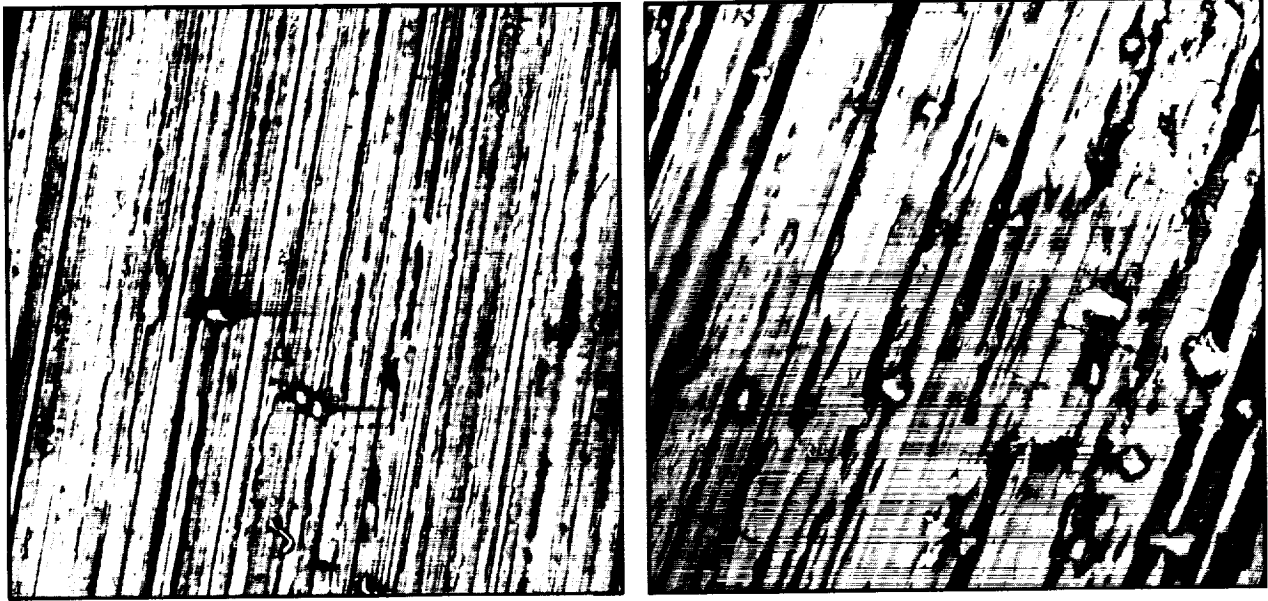


Figure 6. Scanning electron micrographs of surfaces ground under dry condition:
(a) magnification X150, grinding speed 2000 feet/minute.
(b) magnification X150, grinding speed 6000 feet/minute.

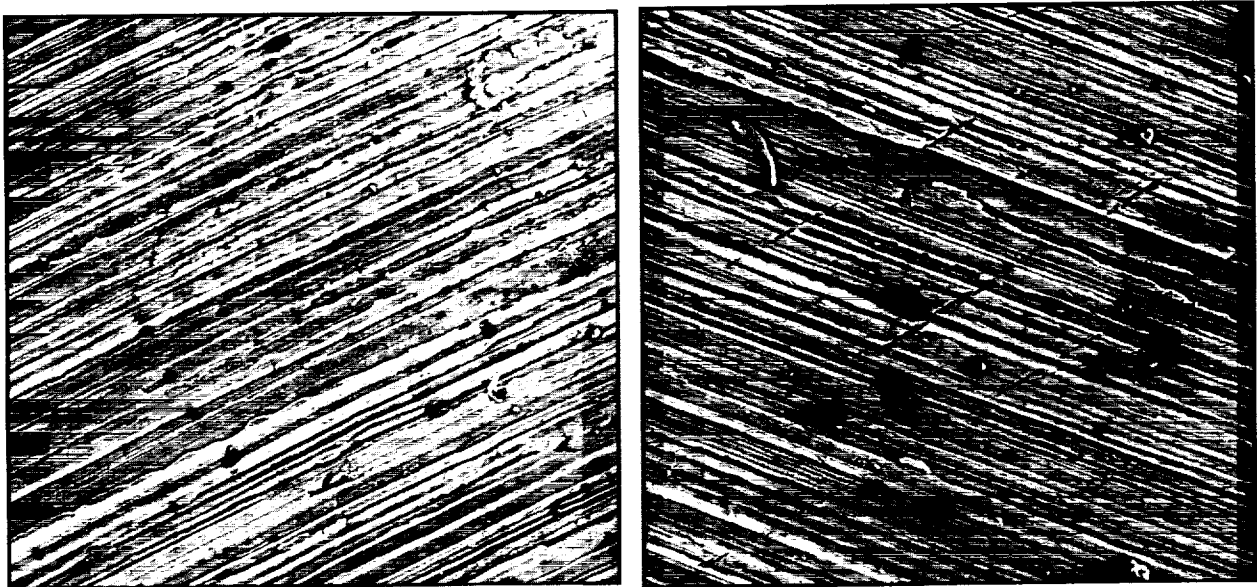


Figure 7. Scanning electron micrographs of surfaces ground under wet condition:
(a) magnification X150, grinding speed 2000 feet/minute.
(b) magnification X150, grinding speed 6000 feet/minute.