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CSC

Lunar Surface Structural Concepts and Construction Studies

Martin Mikulas

Third Annual Symposium
November 21 & 22, 1991

A NASA Space Engineering Research Center at the University of Colorado

LUNAR SURFACE STRUCTURES CONSTRUCTION RESEARCH AREAS

RESEARCH AREA

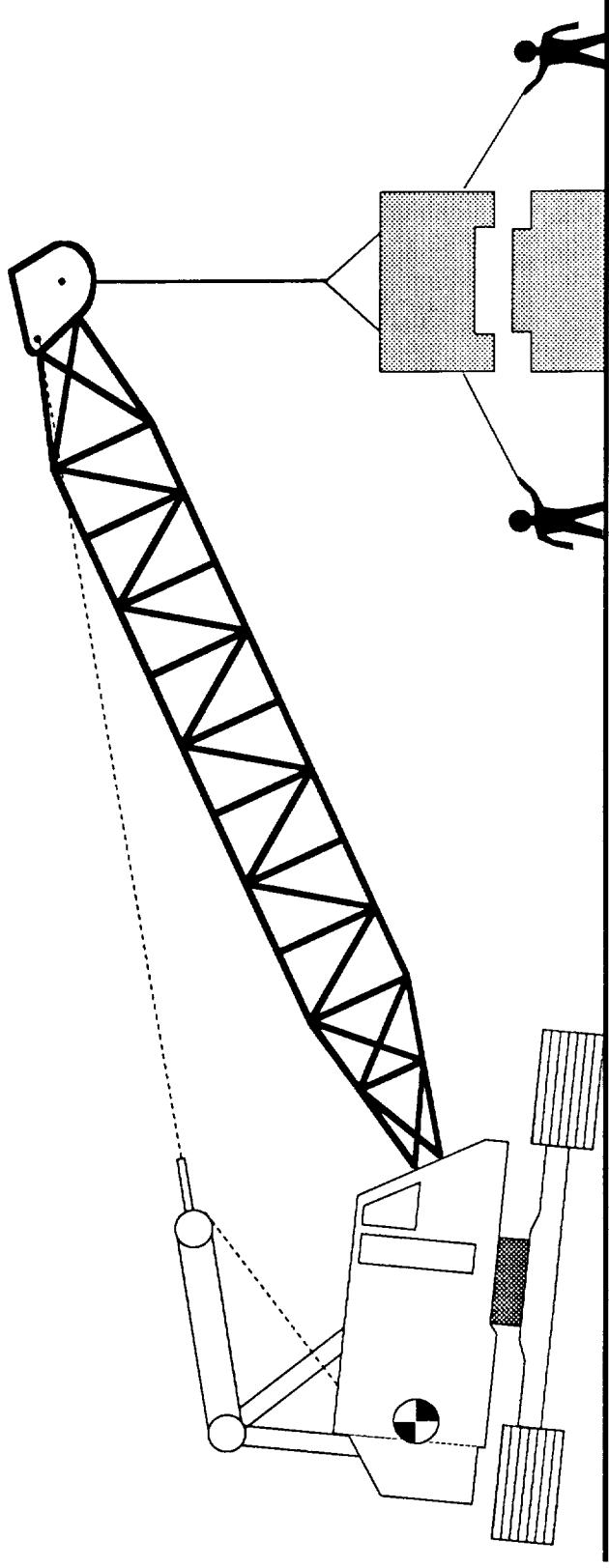
OBJECTIVE

<ul style="list-style-type: none"> - Multiple Cable Crane - Articulating Arm Crane 	<p style="text-align: center;">Remote and/or Precision Positioning Capability For Lunar Construction</p>
<ul style="list-style-type: none"> - Deployable Tower 	<p style="text-align: center;">Automatically Deployable Towers and Beam Type Structures With Minimal Deployment Equipment</p>
<ul style="list-style-type: none"> - Lunar Module Unloading Device 	<p style="text-align: center;">Capability For Self Off-Loading of Modules & Equipment</p>
<ul style="list-style-type: none"> - Deployable Solar Concentrator 	<p style="text-align: center;">Automatically Deployable Reflector With Minimal Deployment Equipment</p>

LUNAR CRANE RELATED DISCIPLINES

- o Remote control and/or autonomous precision construction operations**
- o Multibody dynamics analysis and control of large flexible systems**
- o Analysis and control of cable structures**
- o Quantification of control actuator concepts for large flexible systems**
- o Design of large complex flexible systems**
- o System identification of nonlinear systems**

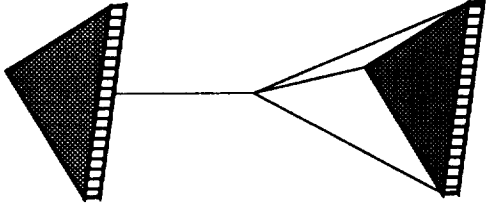
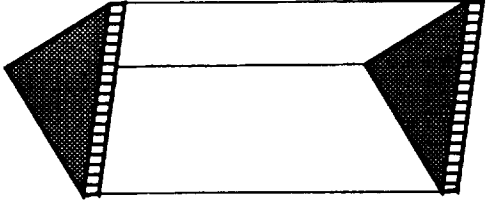
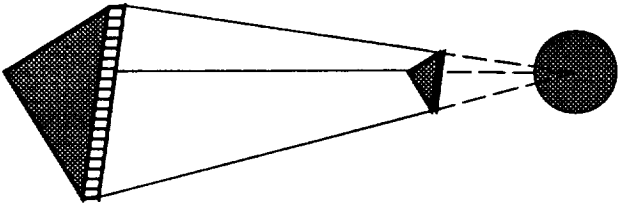
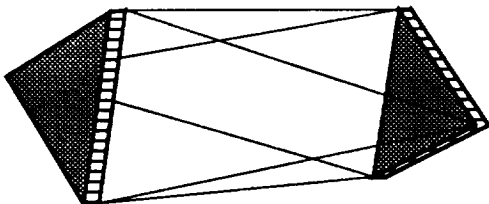
TYPICAL MOBILE CRANE HAS TWO MAJOR SHORTCOMINGS FOR LUNAR BASE APPLICATION



1) Very large mass required to resist tipping

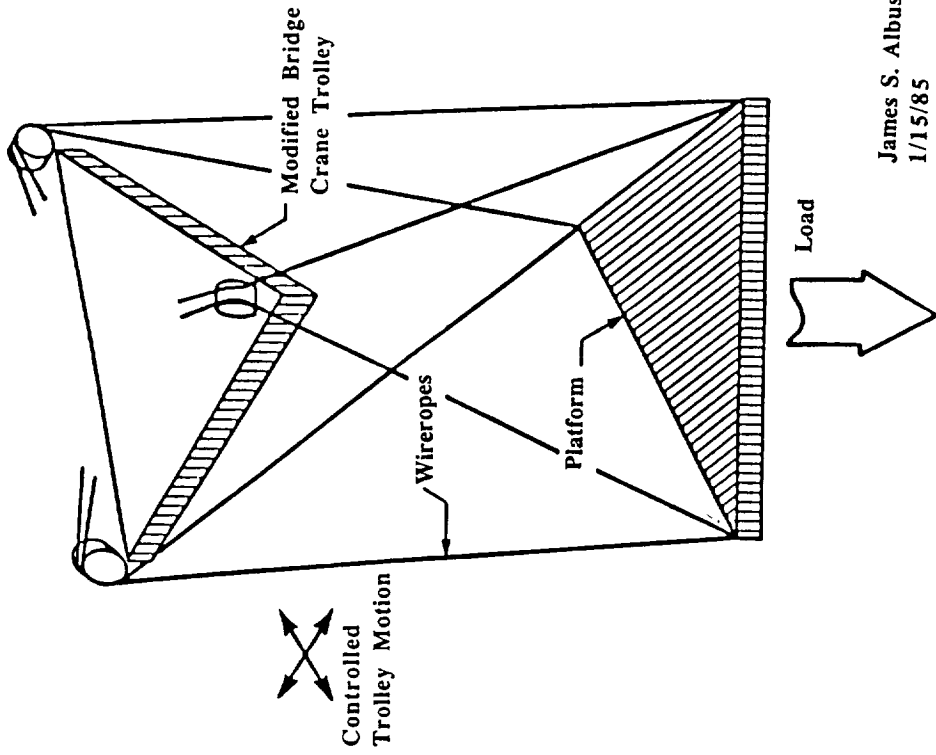
2) Human guidance required for accurate positioning

CANDIDATE CRANE CABLE SUSPENSION SYSTEMS

			
<p>Single Cable</p>	<p>Three Cables</p>	<p>Three Cables</p>	<p>Six Cables</p>
<p>1 DOF Structurally Stiff</p>	<p>3 DOF Structurally Stiff</p>	<p>3 DOF Structurally Stiff 3 DOF Stiffened by Triangulated Cables</p>	<p>6 DOF Structurally Stiff</p>

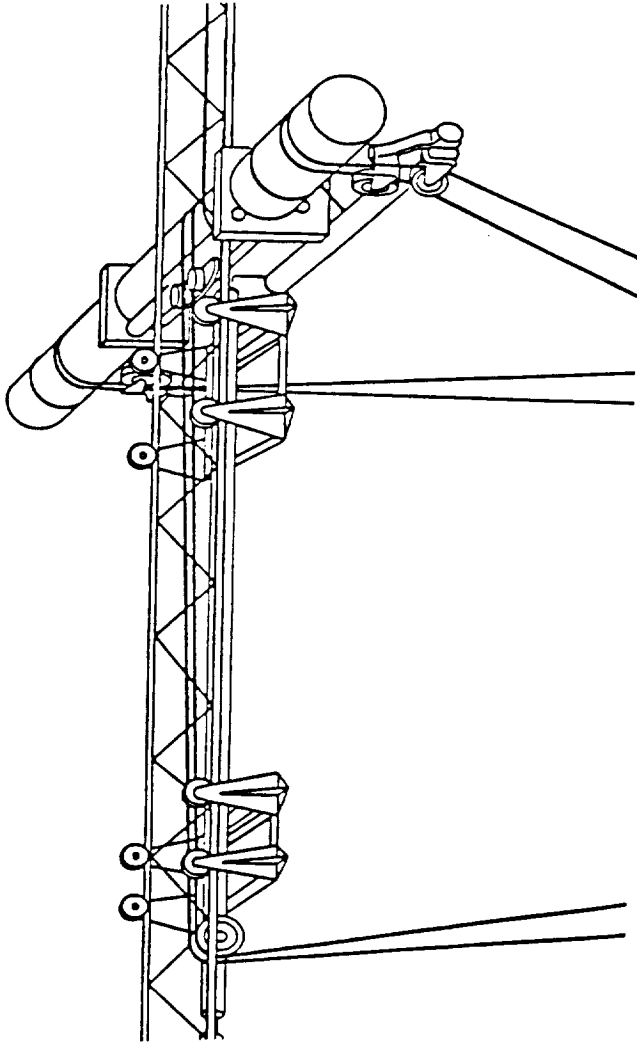
NIST SIX-CABLE SUSPENSION CRANE

Cable Geometry



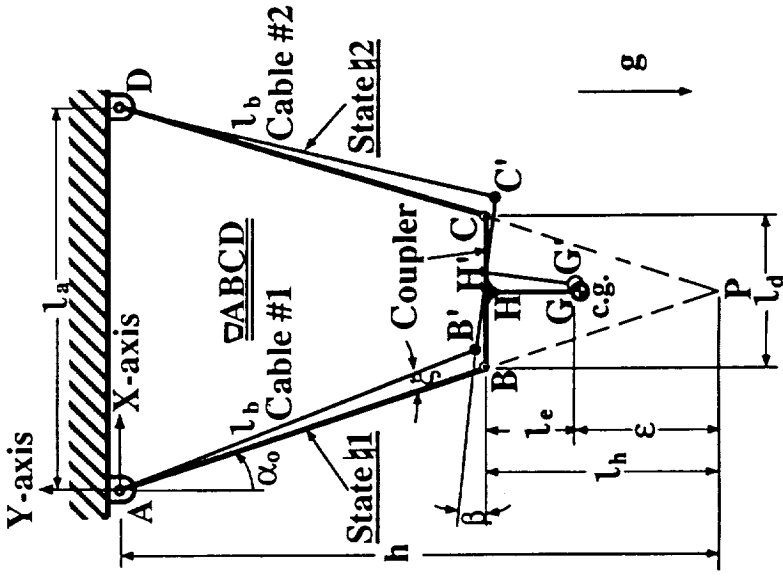
James S. Albus
1/15/85

Cable Drive System

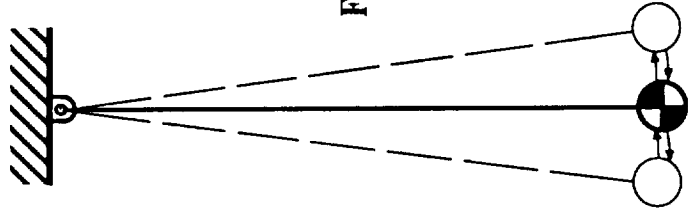


NUMERICAL EXAMPLE OF NATURAL FREQUENCY

A Symmetric Model

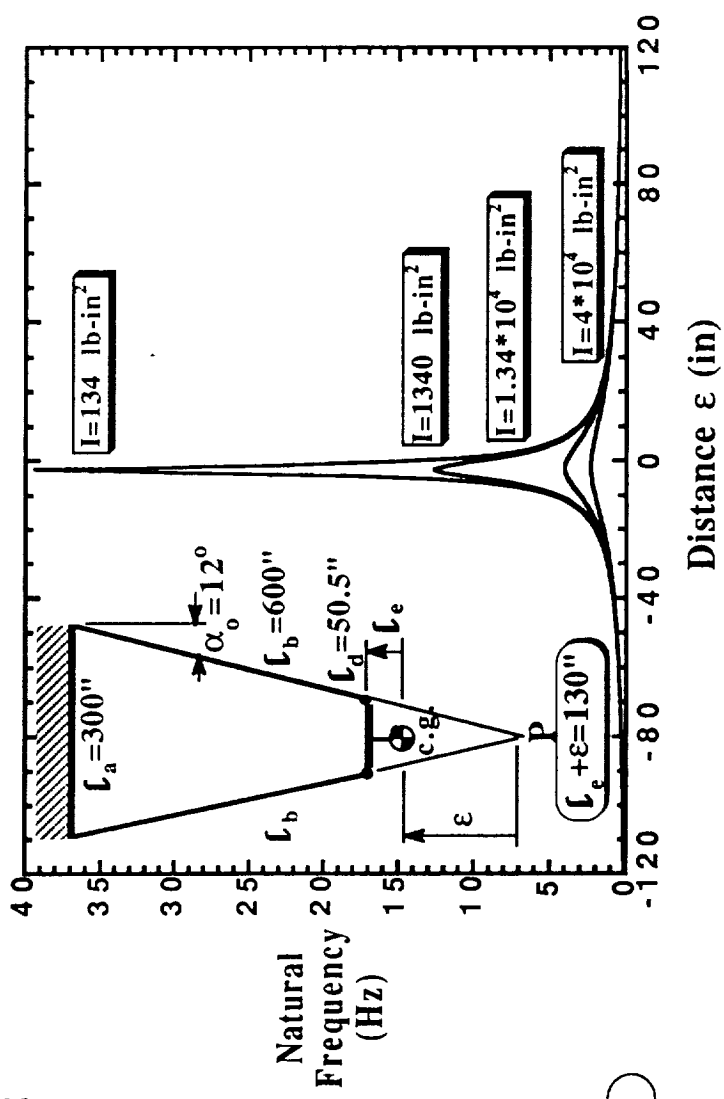


A Swinging Pendulum

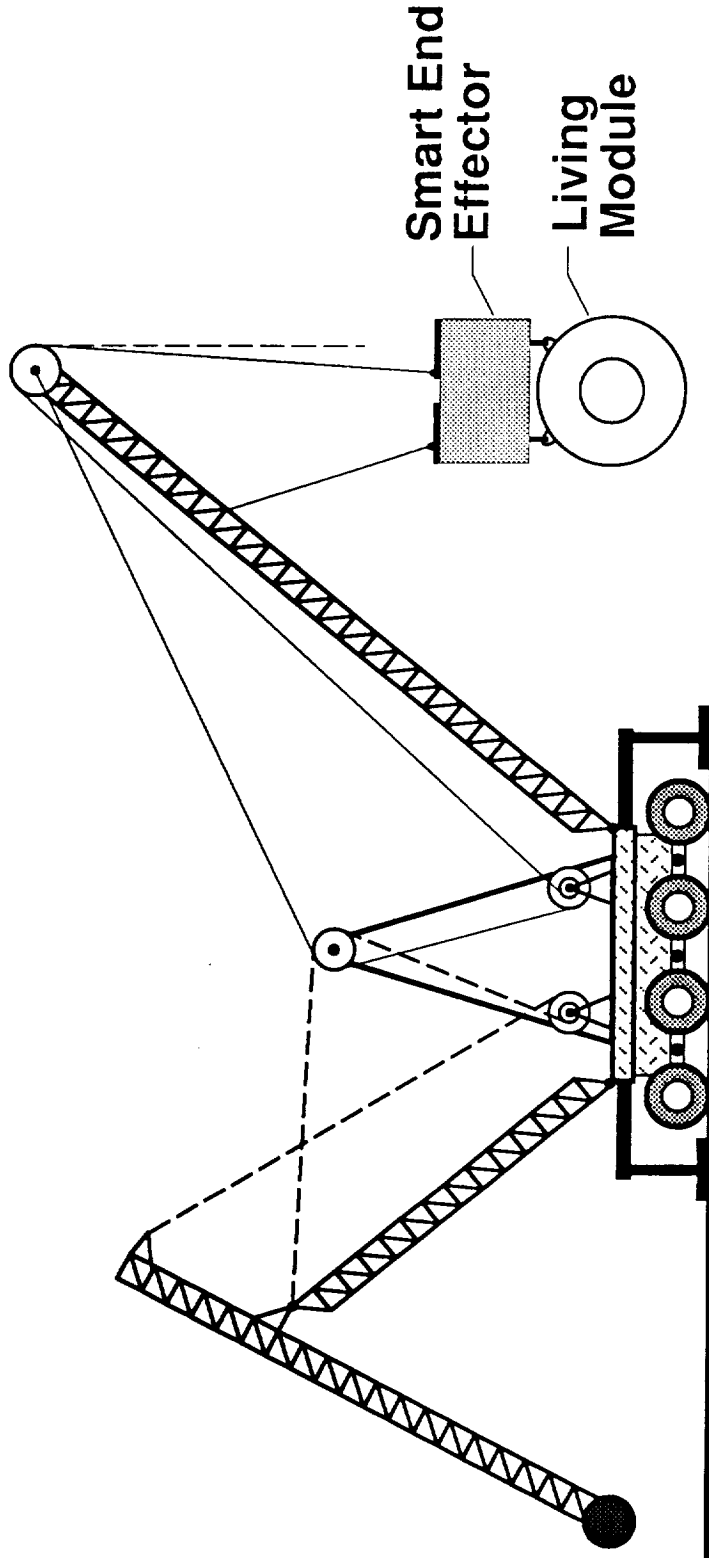


$$F = \sqrt{\frac{\left(\frac{l_h}{h-l_h}\right) \left[l_h h + \frac{l_a^2}{4} \right] + l_e h}{\epsilon^2 + \rho^2}} F_{pendulum}$$

$$F_{pendulum} = \frac{1}{2\pi} \sqrt{\frac{g}{h}}$$



COUNTER-BALANCED ACTIVELY-CONTROLLED LUNAR CRANE INCORPORATES TWO NEW FEATURES FOR IMPROVED PERFORMANCE



1) Active Counter Weight to Reduce Overturning Moment

2) Multiple Payload Suspension Cables to Provide Stable Precision Positioning

LUNAR CRANE PENDULUM MECHANICS

3 Translations Have Structural Stiffness

3 Rotations Have Pendulum Stiffness

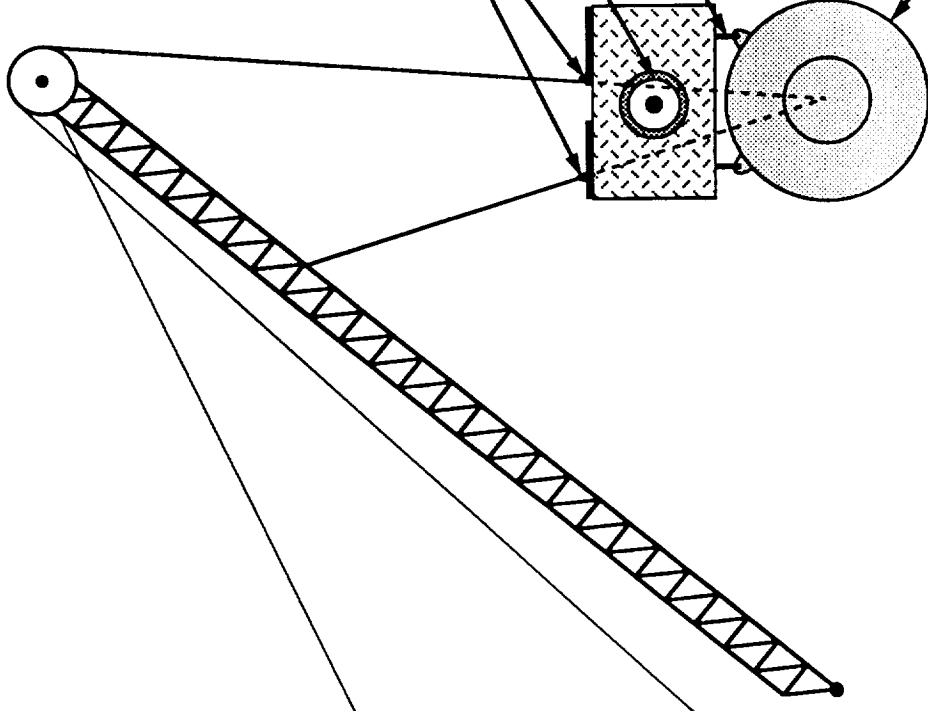
Potential Control Mechanisms

Active Cable Positioners

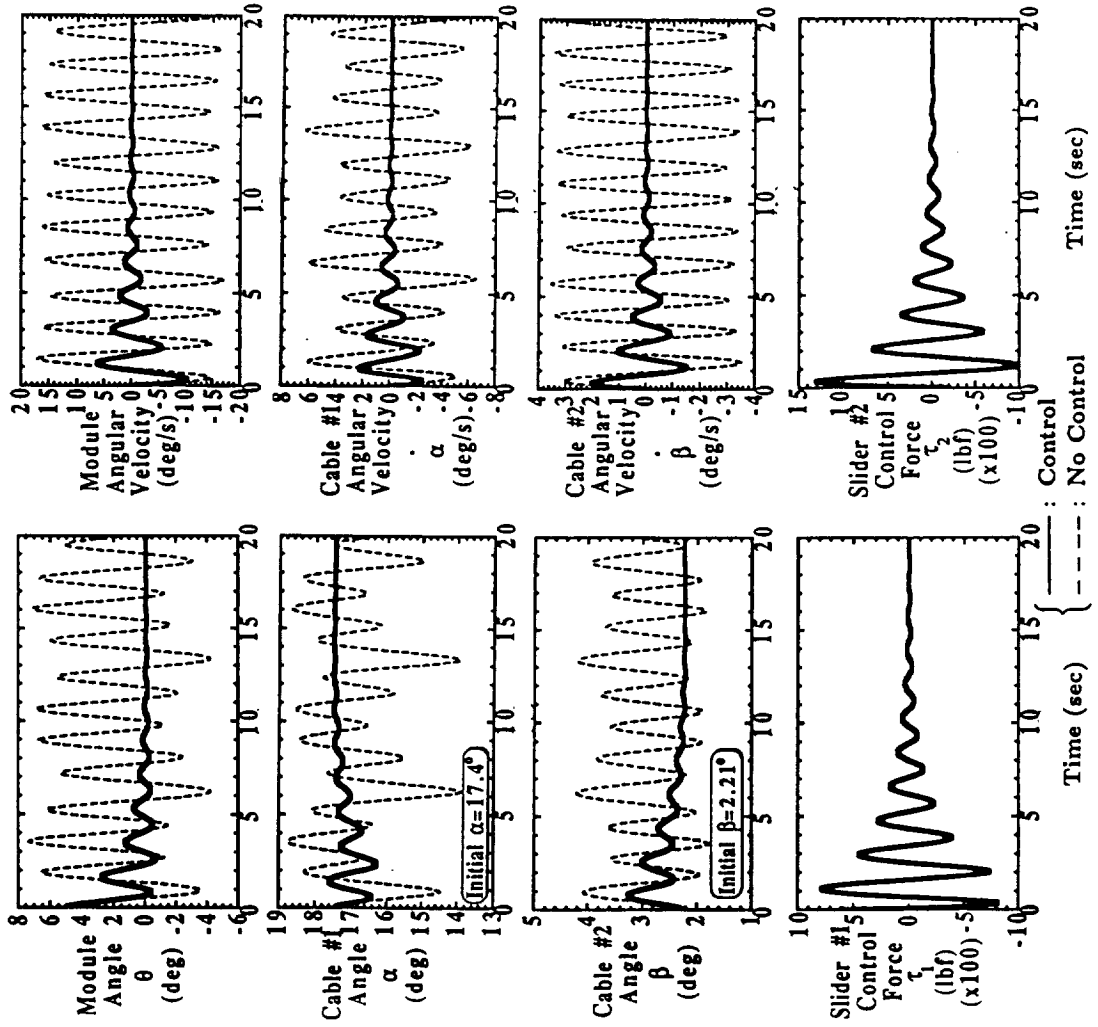
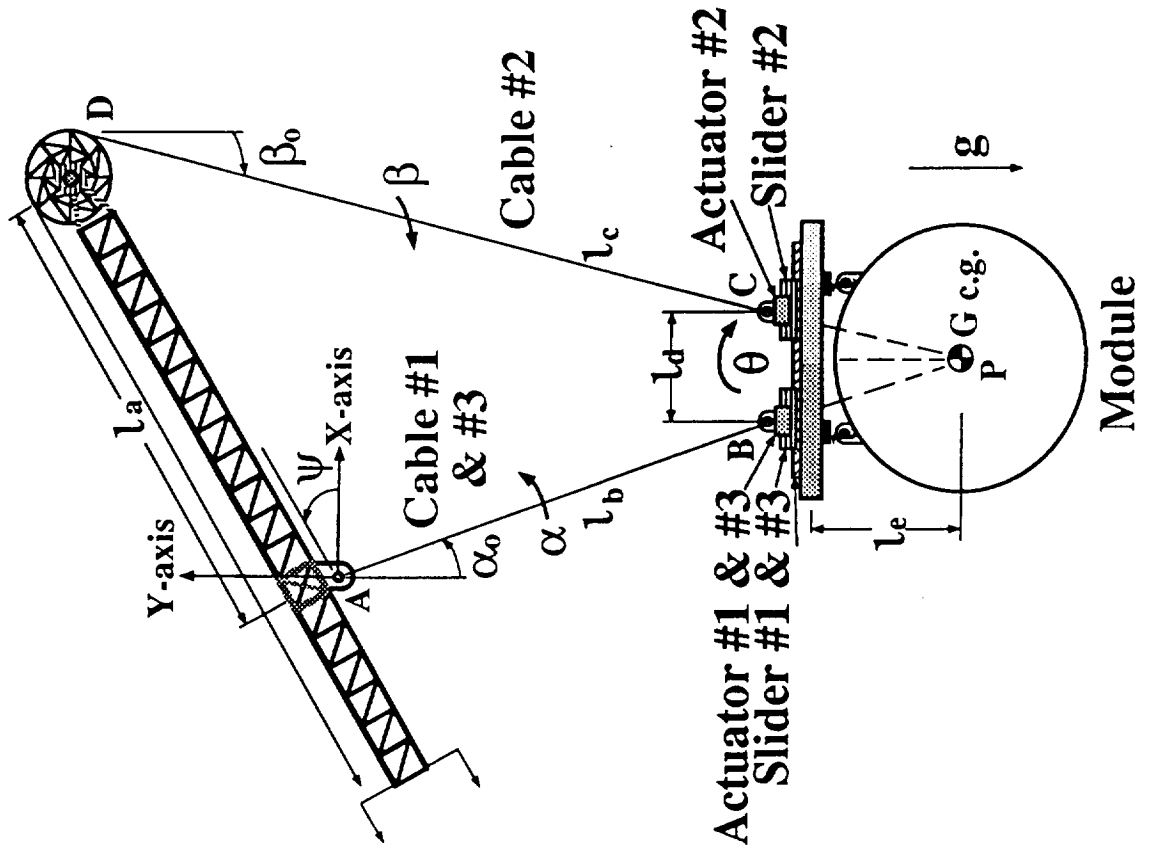
Active Inertia Wheels

Active Attachments

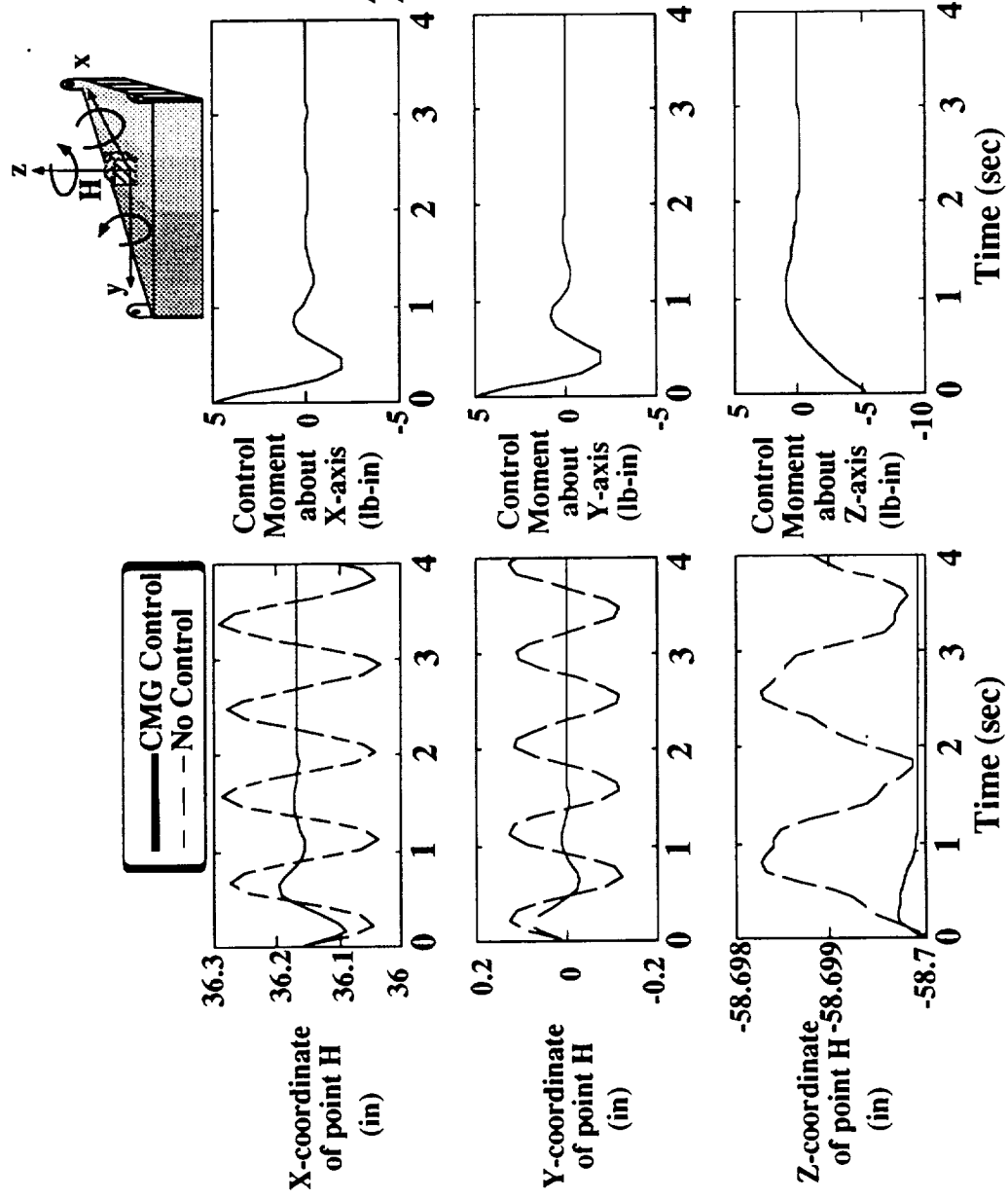
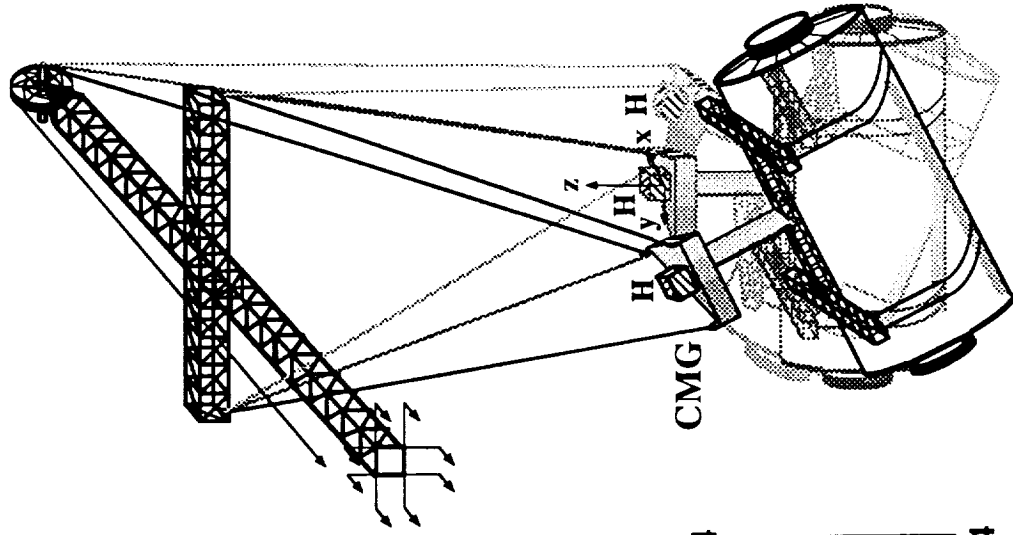
Payload (M, I)



SIMULATION RESULTS (II)

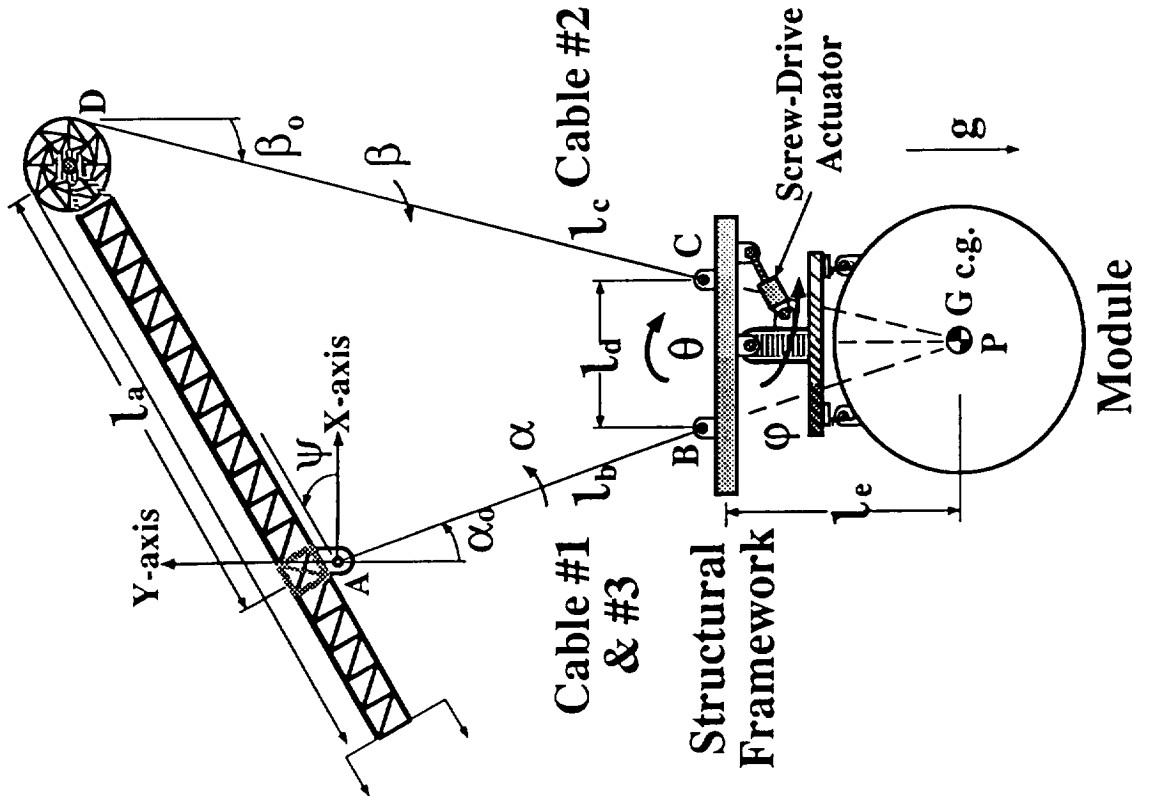
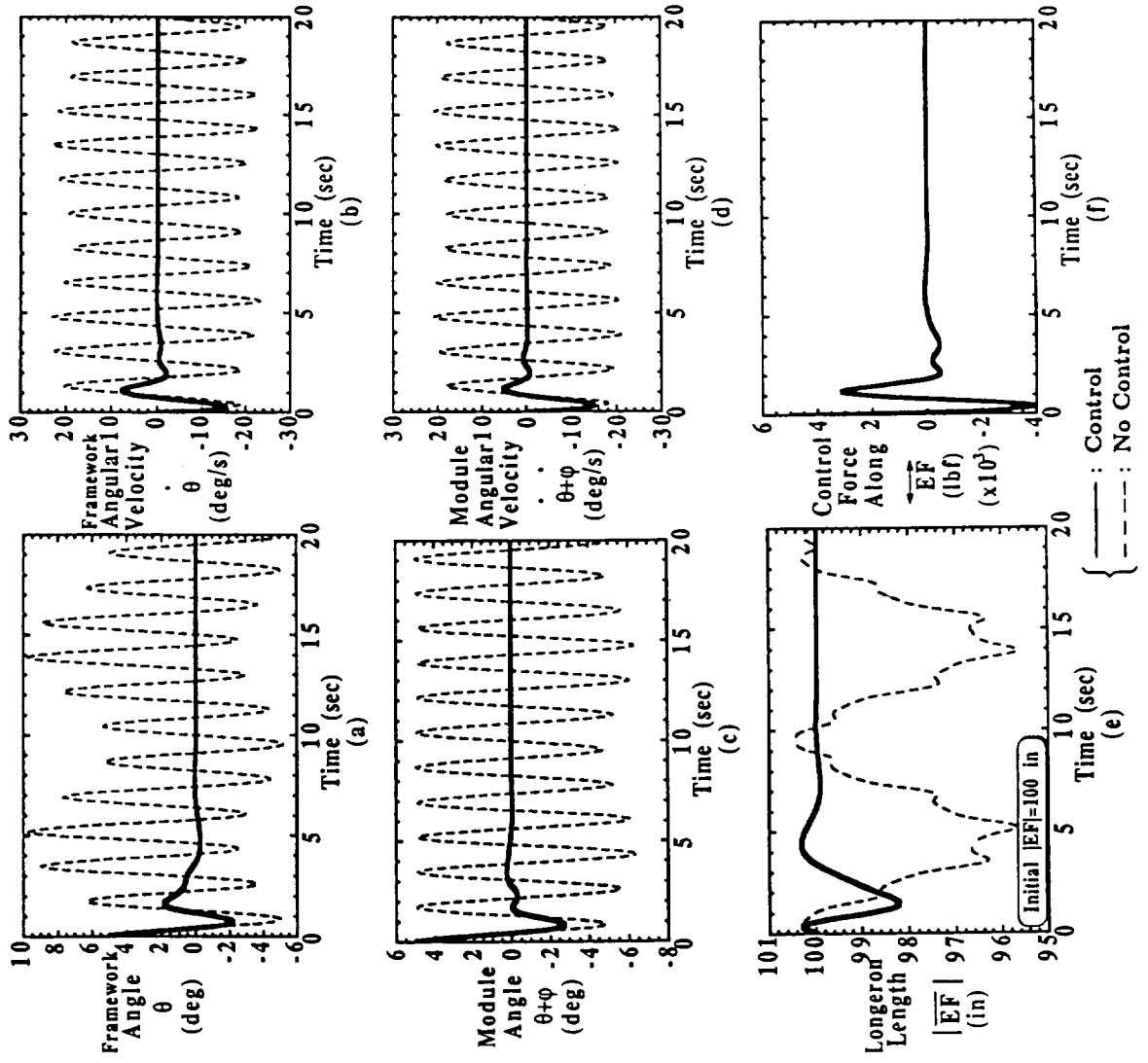


CMG CONTROL SIMULATION RESULTS

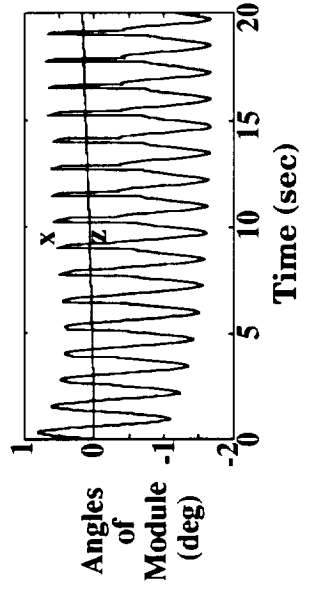
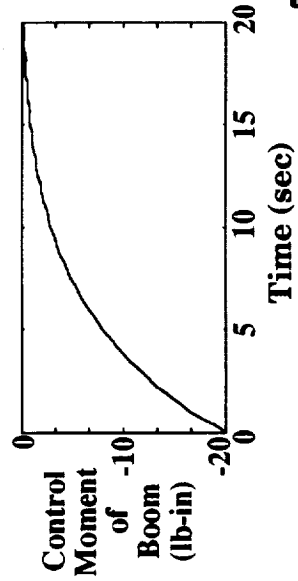
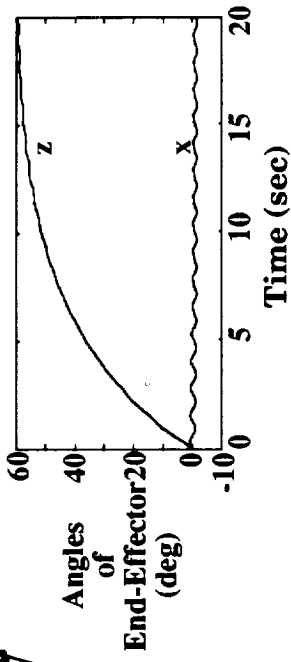
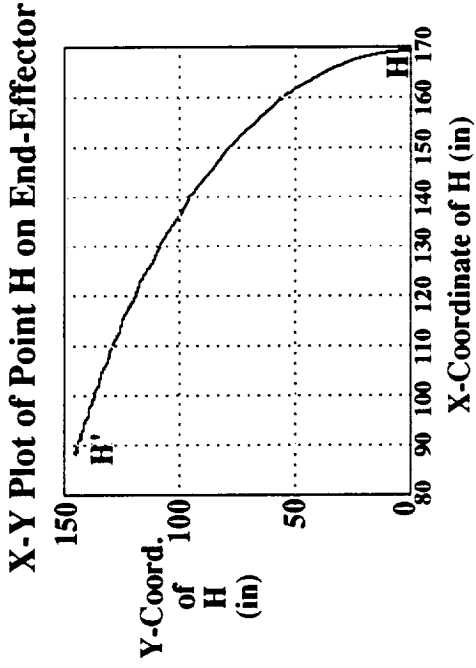
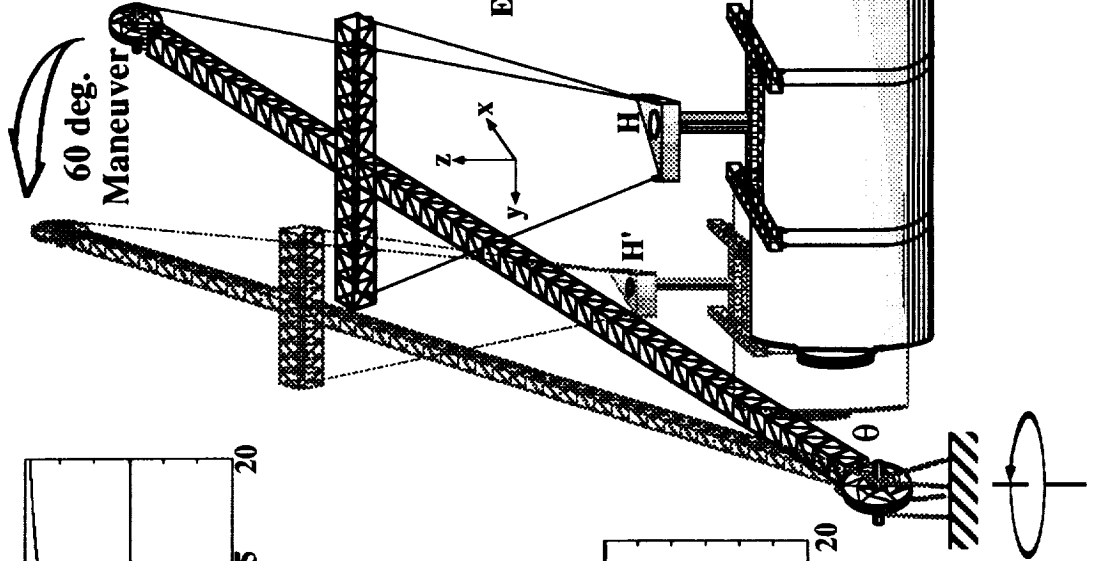
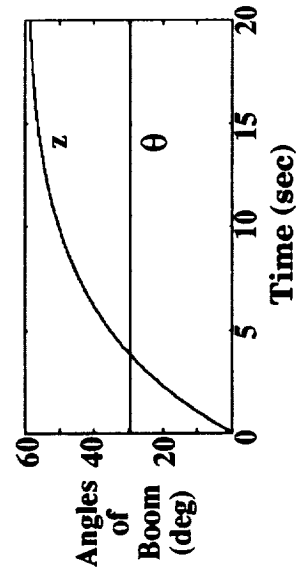


SIMULATION RESULTS

(I)

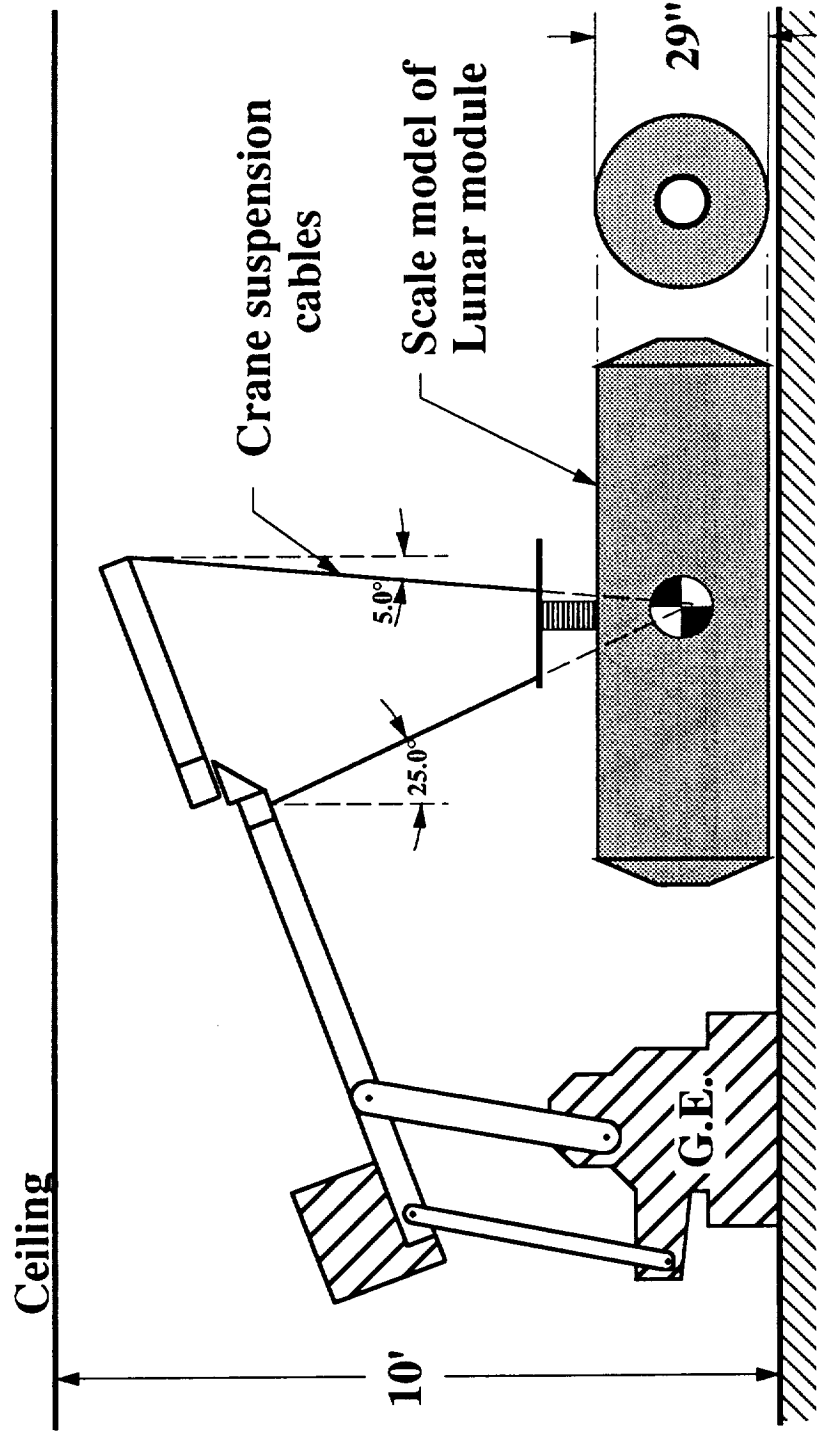


SLEWING SIMULATION RESULTS

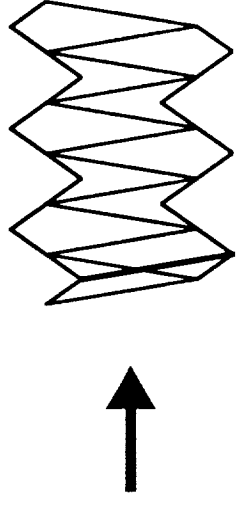
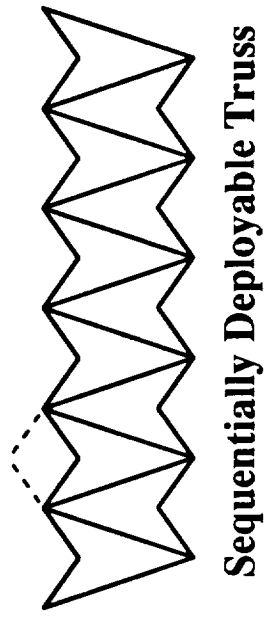
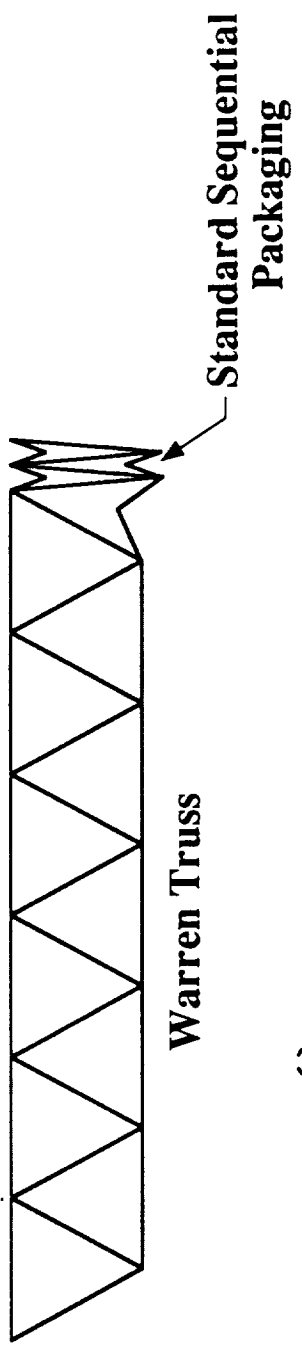


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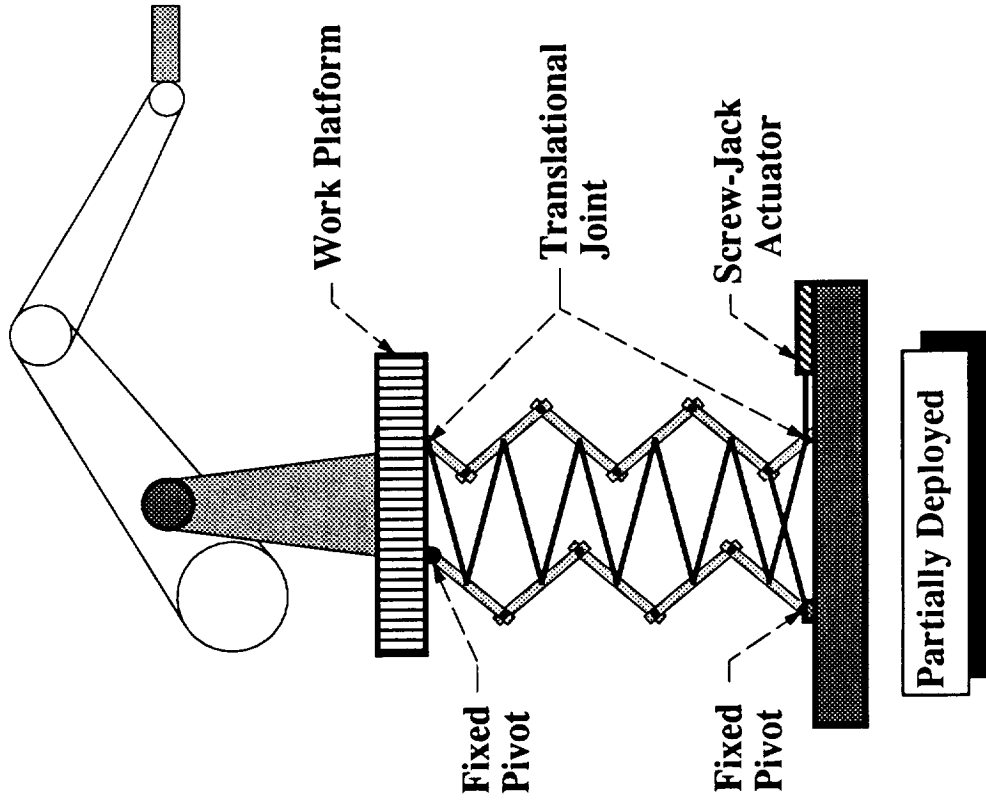
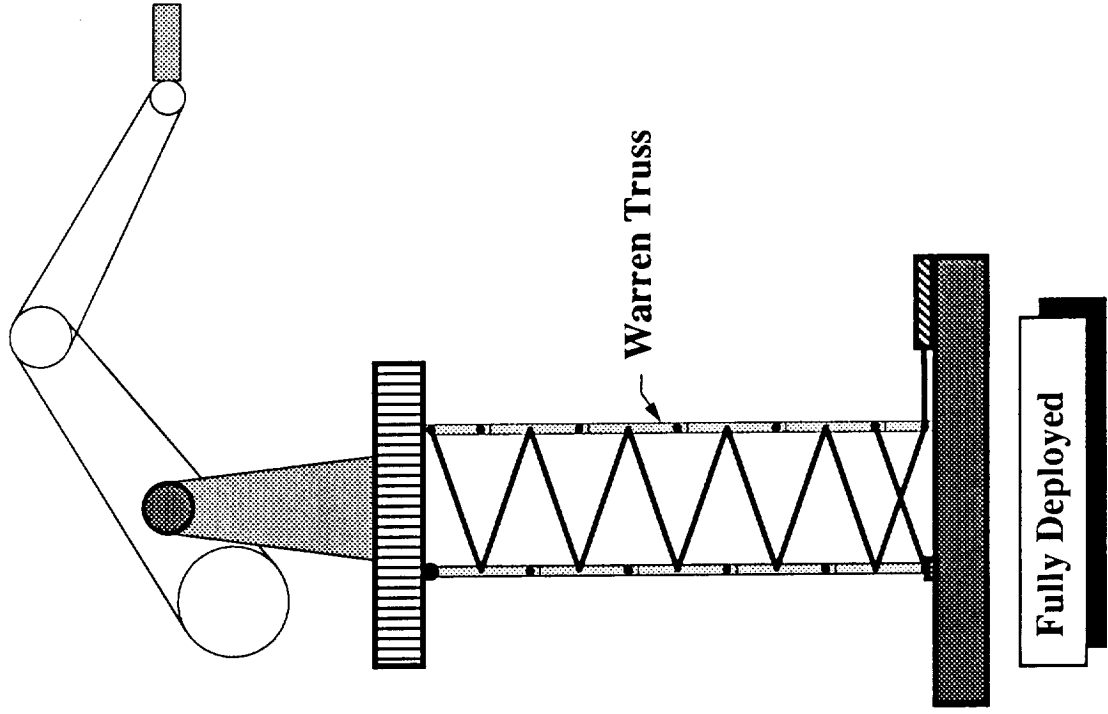
ONE-SIXTH SCALE LUNAR CRANE TEST-BED USING G.E. ROBOT FOR GLOBAL MANIPULATION.



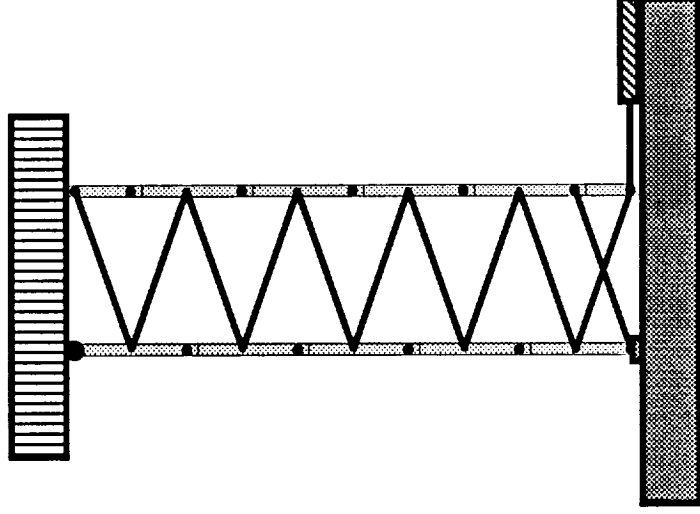
BASIC DEPLOYABLE TRUSS APPROACHES



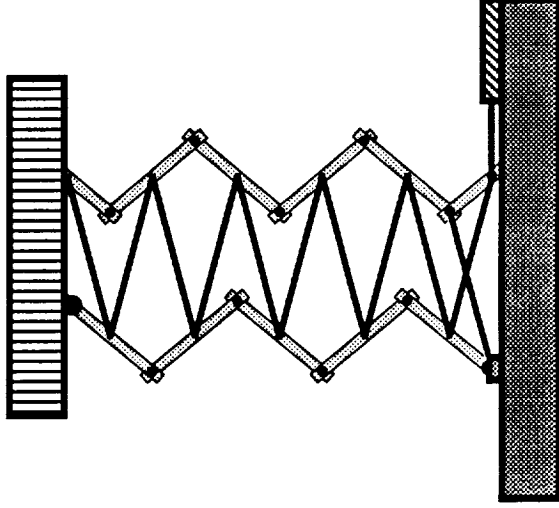
BI-PANTOGRAPH ELEVATOR PLATFORM



COMPARISON OF ELEVATOR PLATFORMS

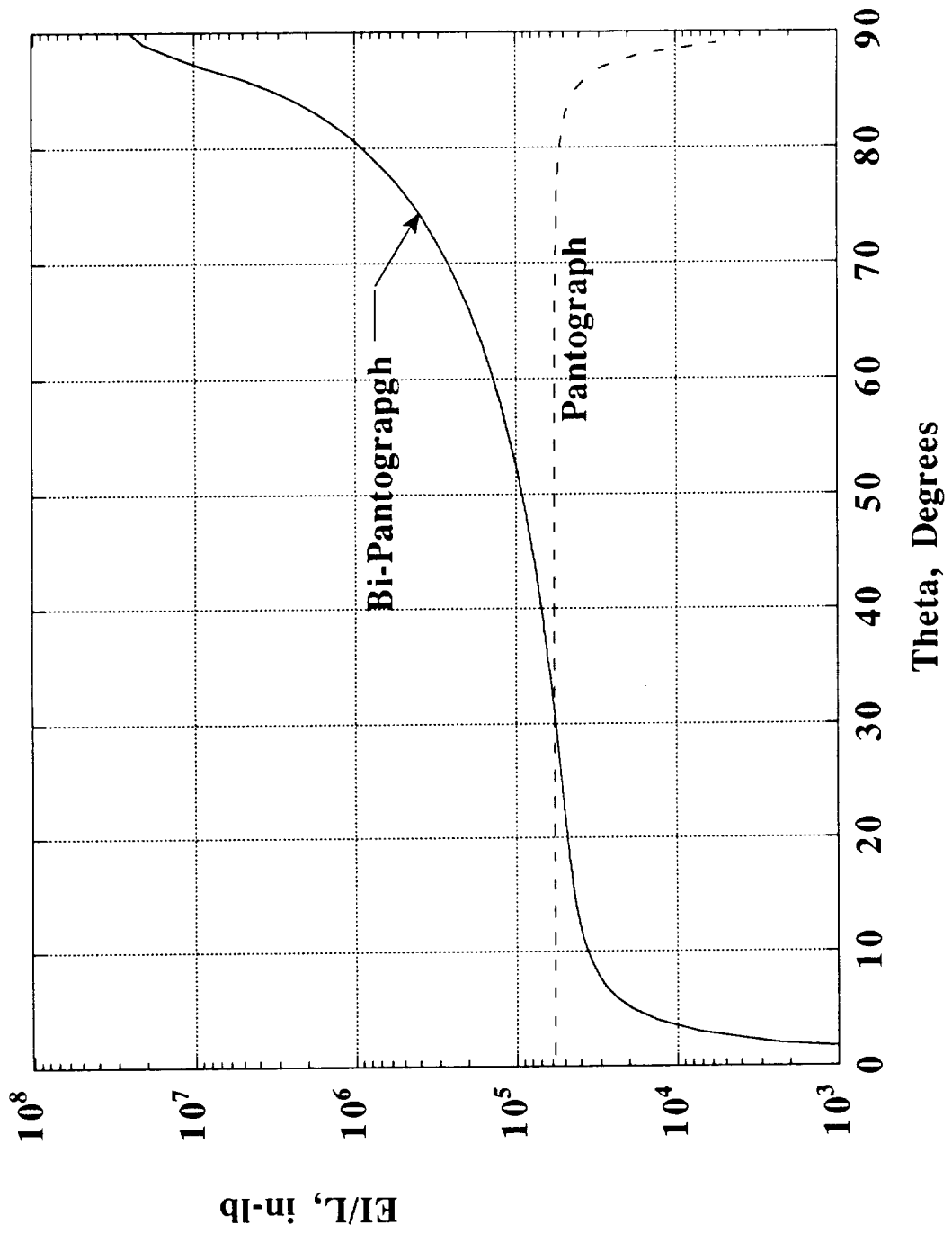


Bi-Pantograph

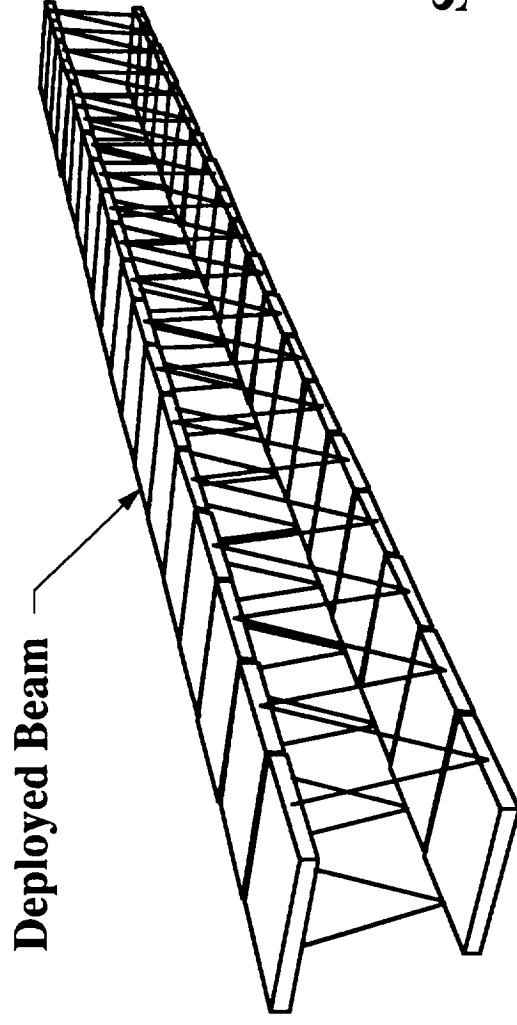


Pantograph

BI-PANTOGRAPH VS PANTOGRAPH STIFFNESS

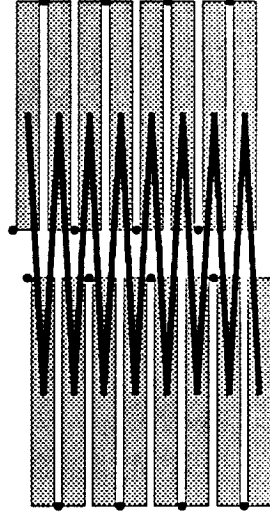


PERSPECTIVE OF BI-PANTOGRAPH BEAM



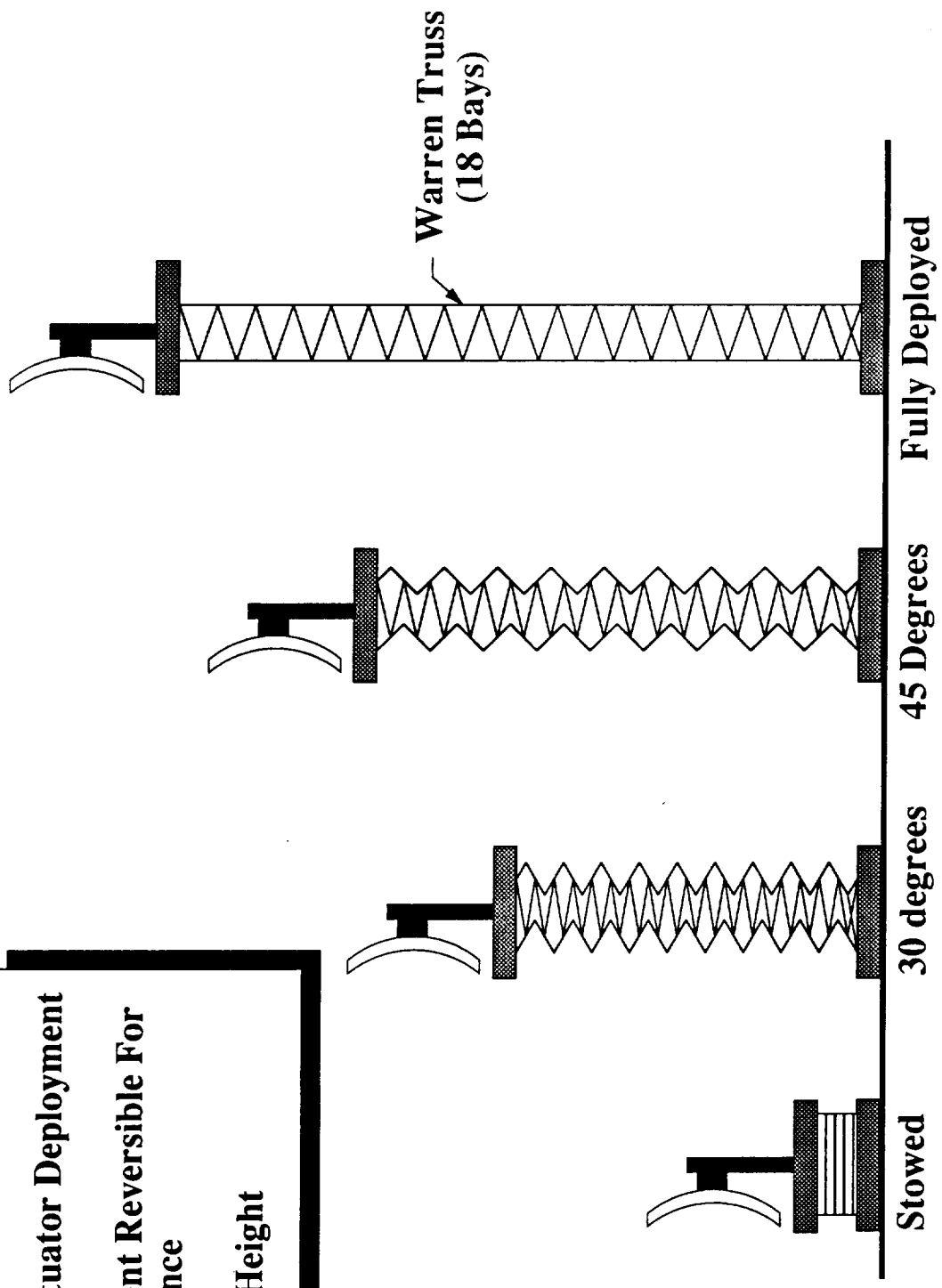
Deployed Beam

Stowed Beam

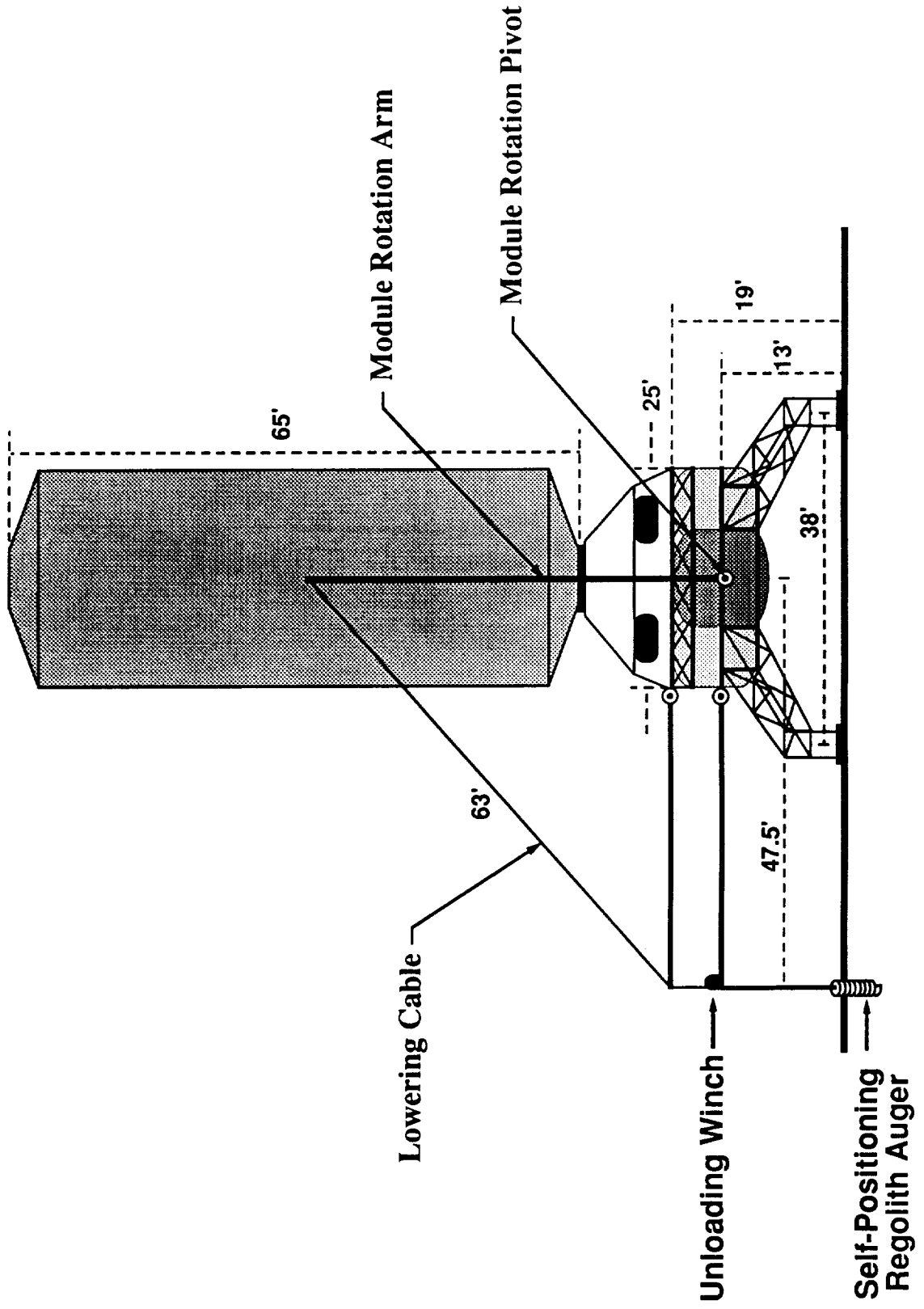


BI-PANTOGRAPH SYNCHRONOUSLY DEPLOYABLE TOWER/BEAM

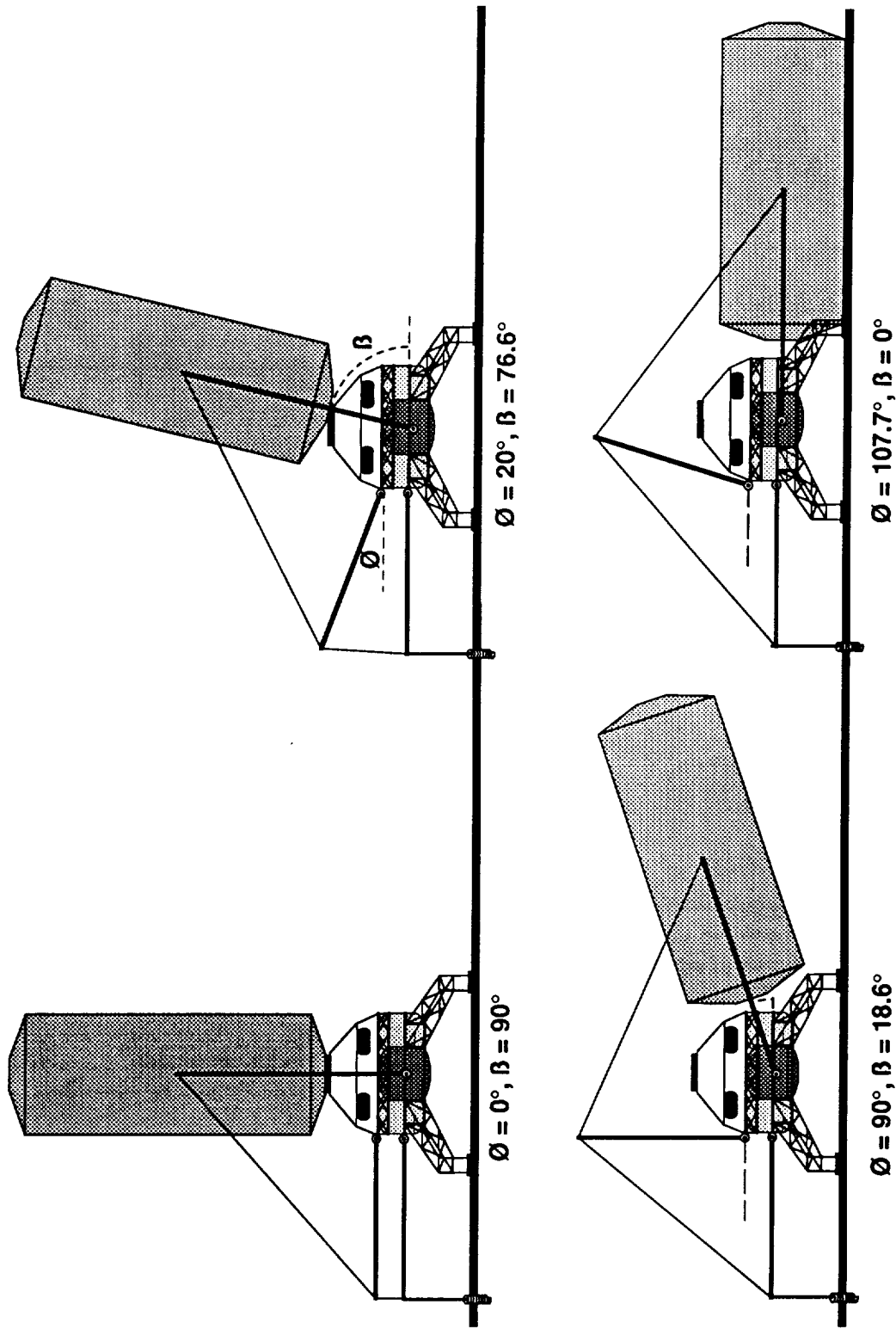
- Single Actuator Deployment
- Deployment Reversible For Maintenance
- Variable Height



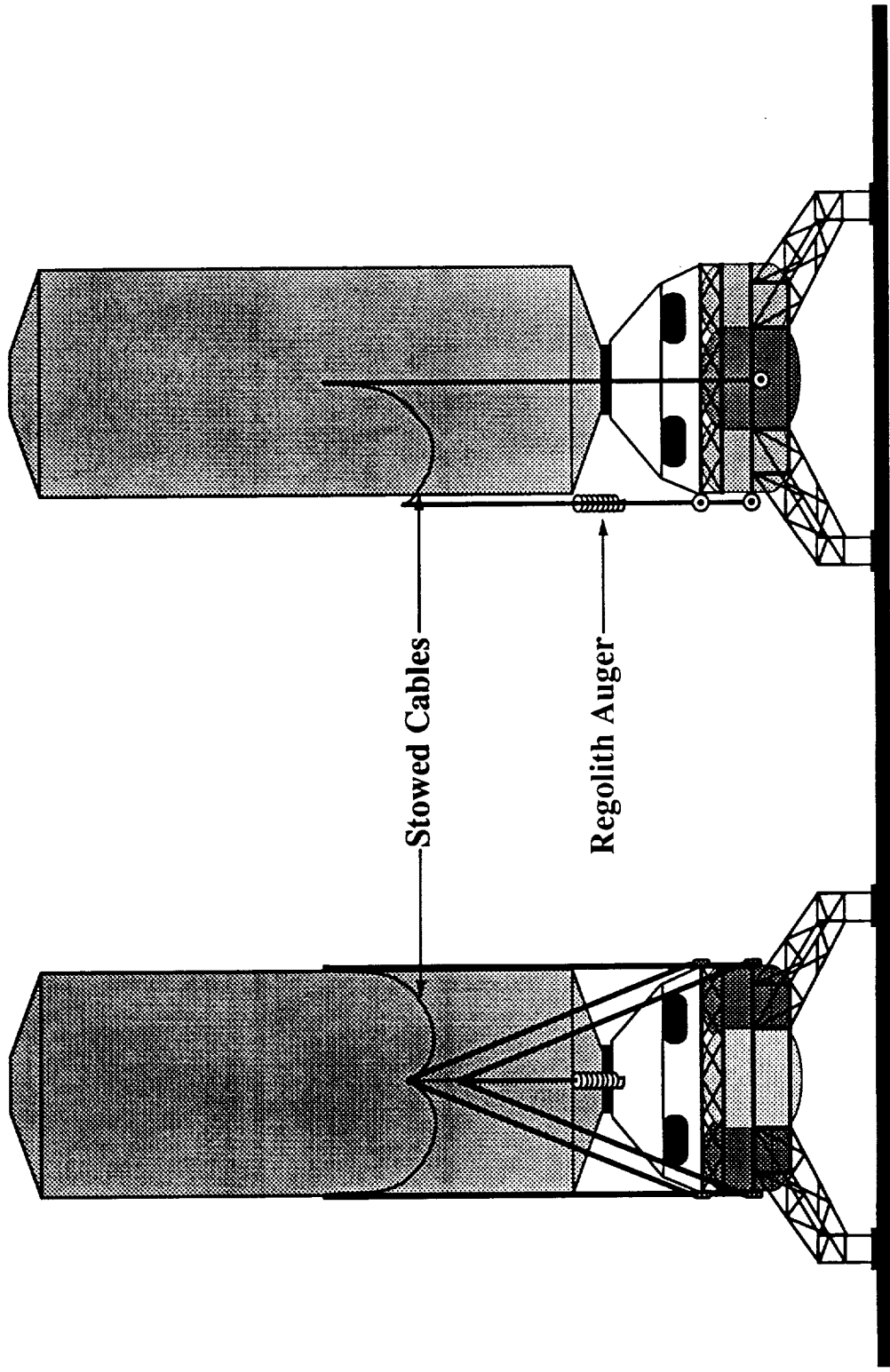
LUNAR MODULE OFF-LOADING CONCEPT



LUNAR MODULE OFF-LOADER CONCEPT DURING VARIOUS PHASES OF OPERATION



MODULE OFF-LOADER CONCEPT PACKAGED (REAR & SIDE VIEWS)



STARBURST DEPLOYABLE PRECISION REFLECTOR

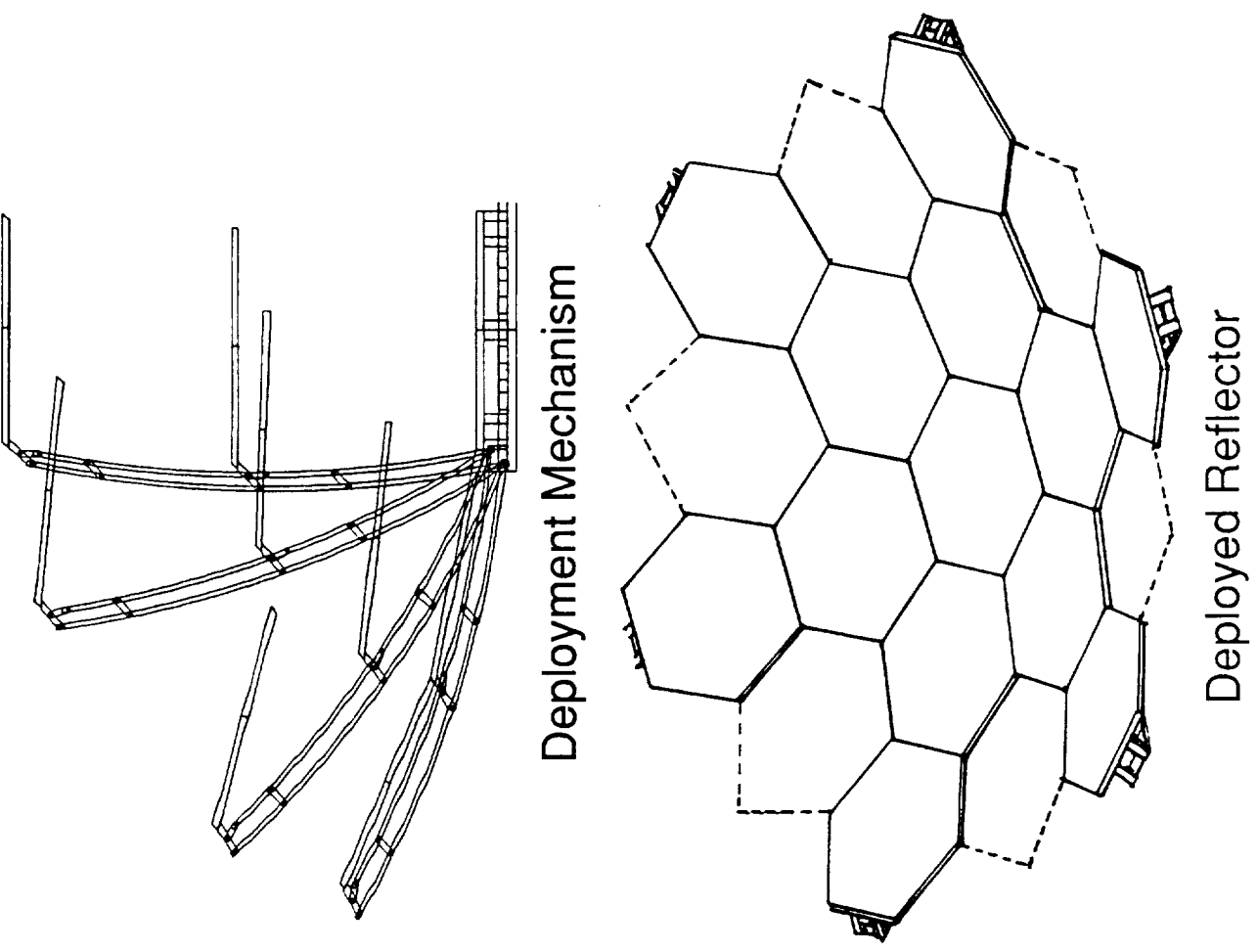
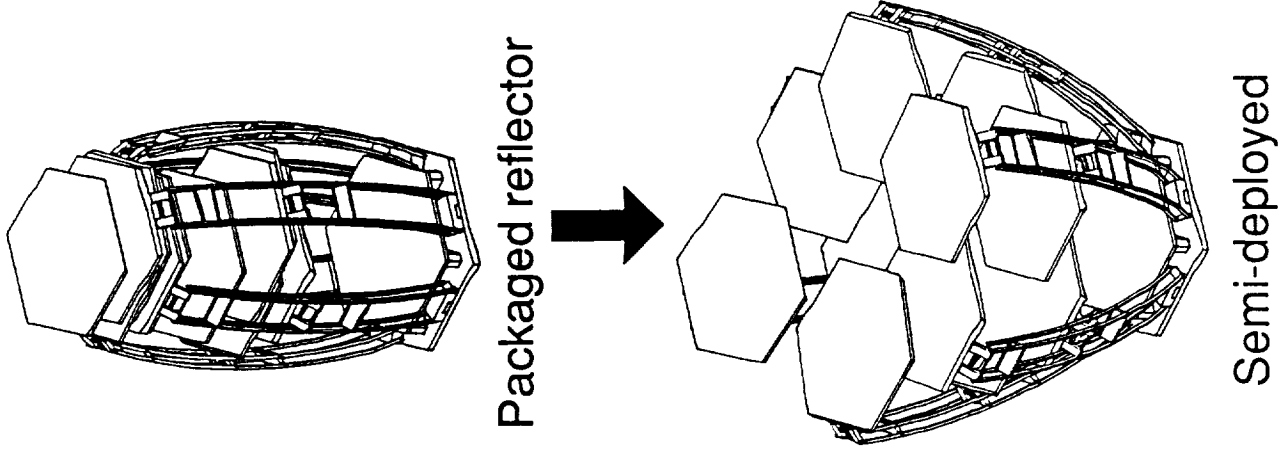
Features

- Maximum packaging efficiency for reflector panels
- Simple one-degree-of-freedom deployment of reflector arms
- Permits integrated reflector system

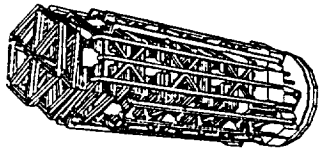
Applications

- LDR-type telescopes
- Microwave radiometers
- Solar concentrators

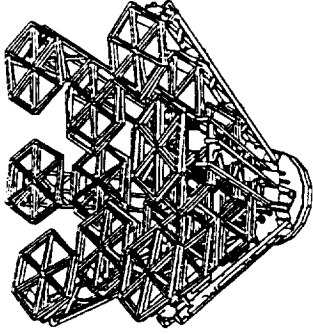
“STAR BURST” CONCEPT HAS POTENTIAL FOR DEPLOYING 20 METER DIAMETER PRECISION DEFLECTOR



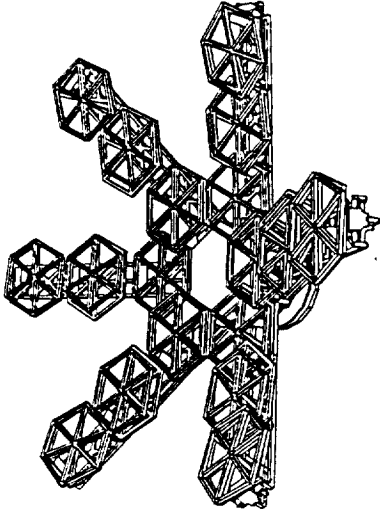
STARBURST DEPLOYABLE PRECISION REFLECTOR



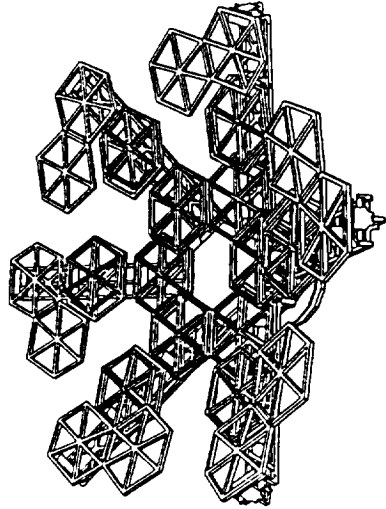
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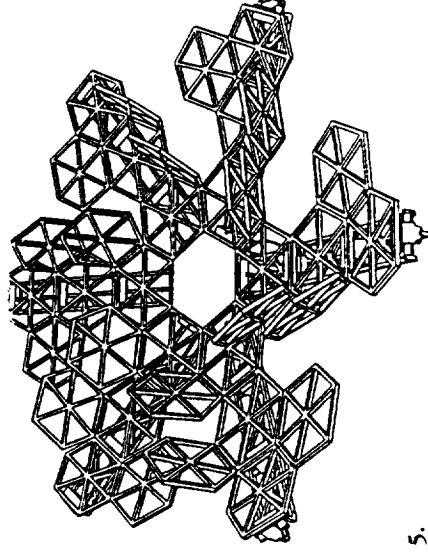
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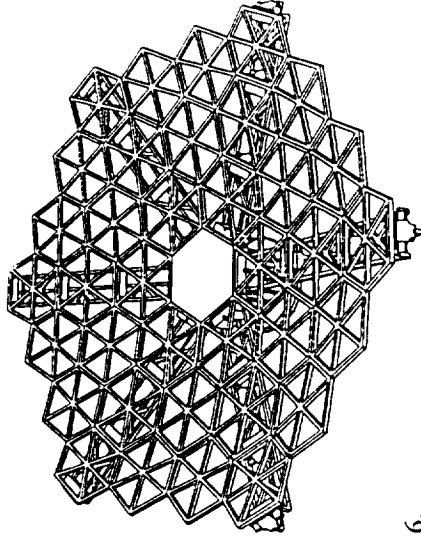
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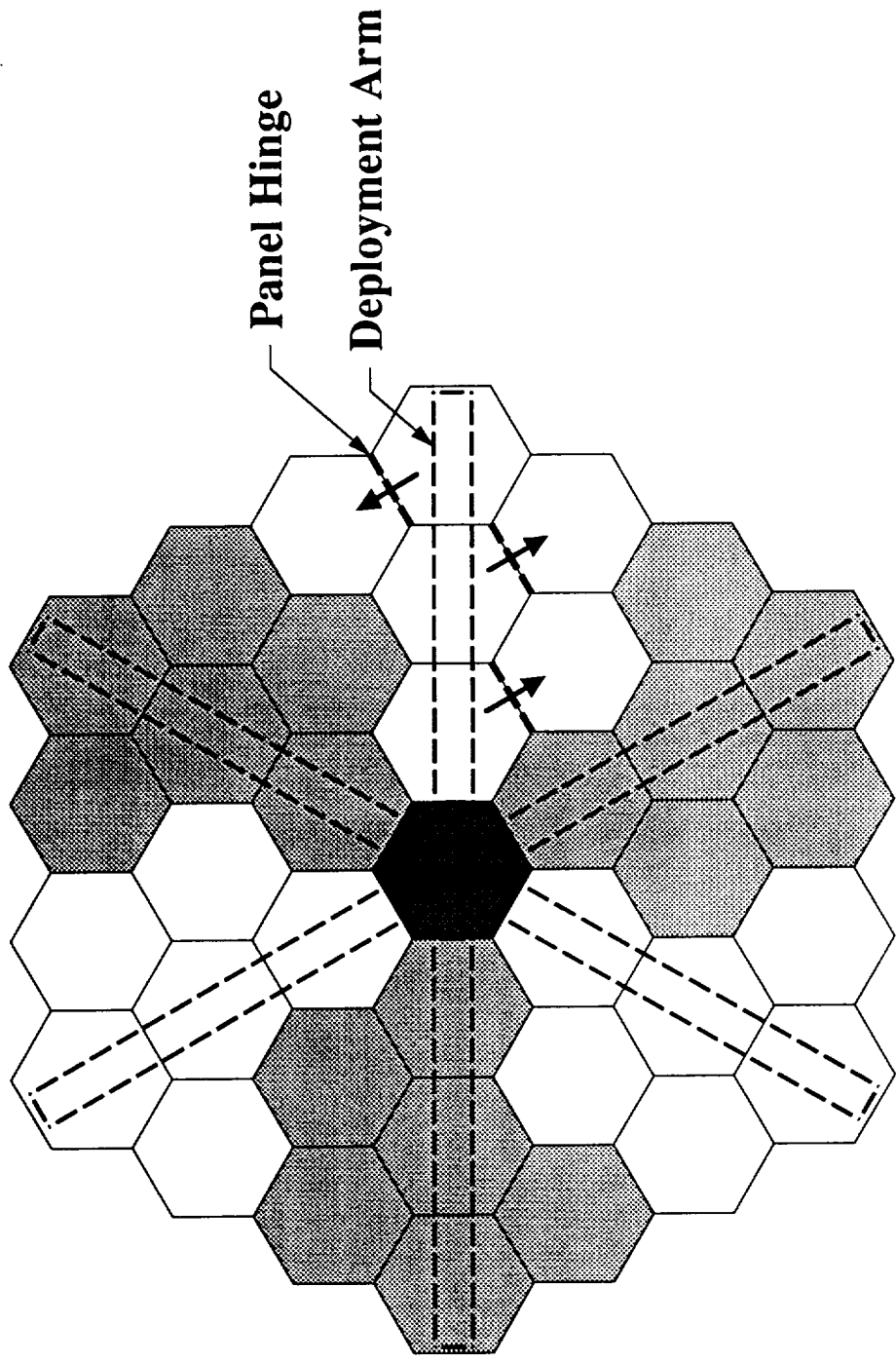
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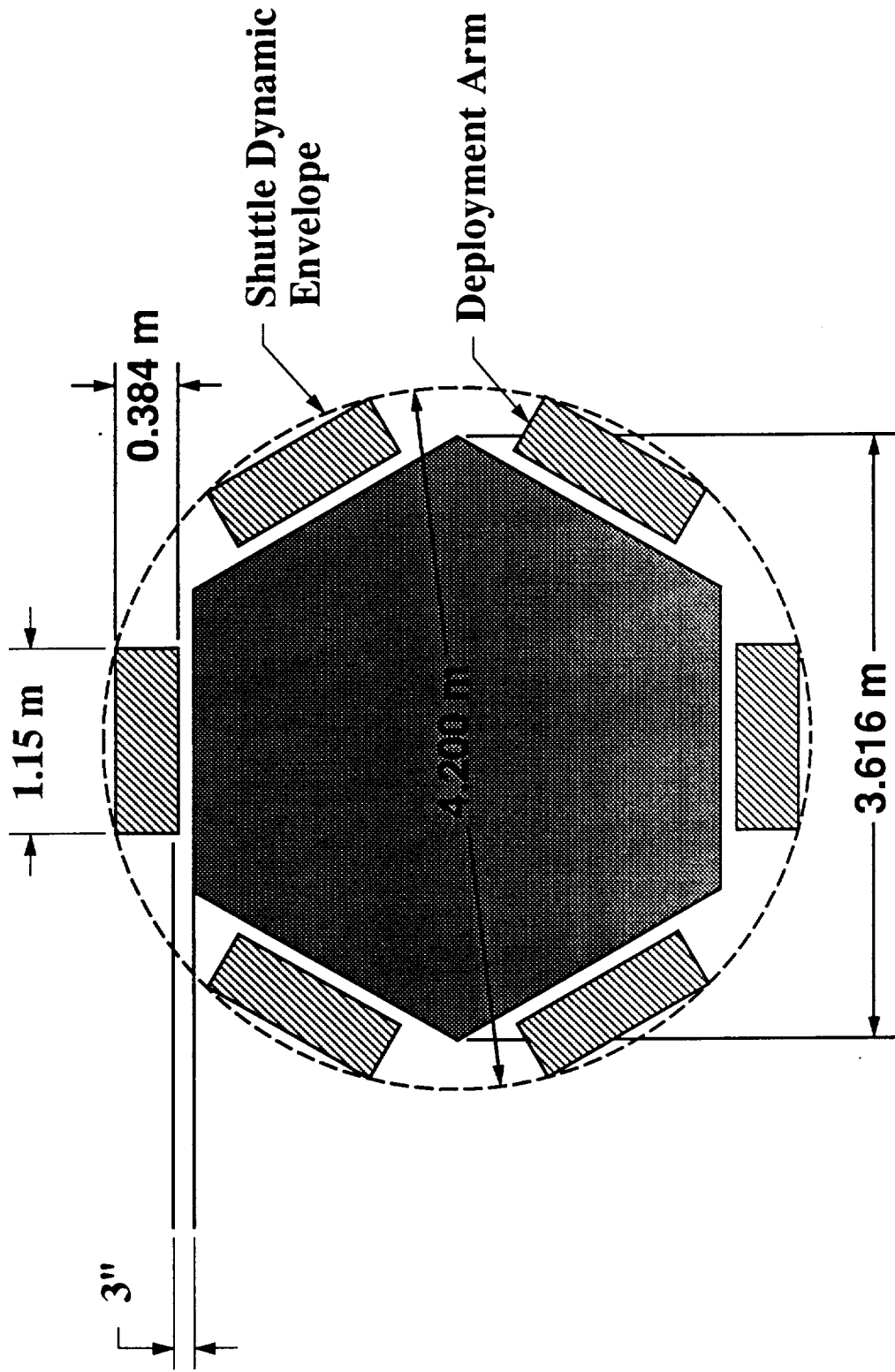
6.

3 RING REFLECTOR DEPLOYMENT SCHEME

- 37 Panels Total
- 6 Deployment Arms
- 6 Panels Per Deployment Arm

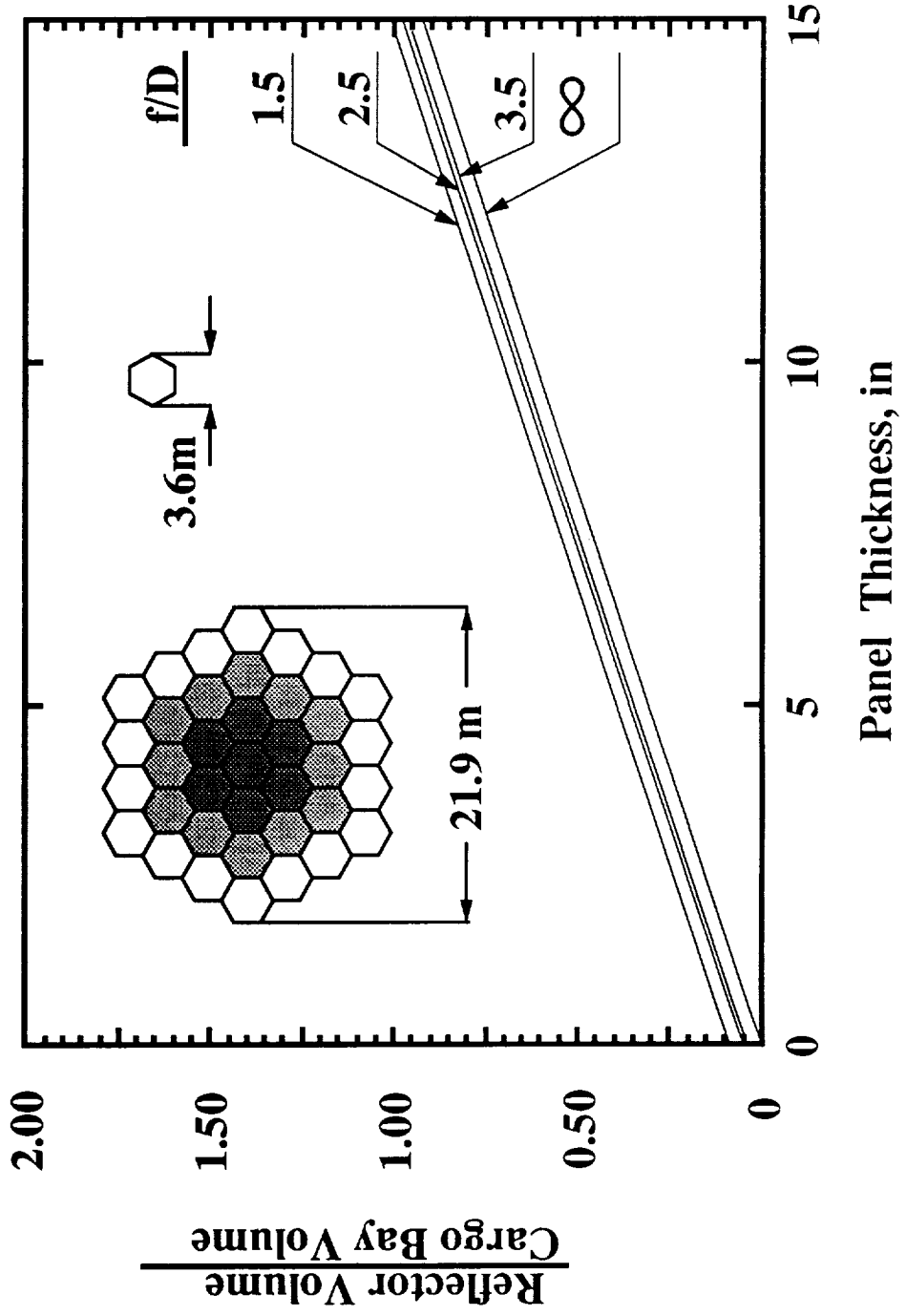


CROSS-SECTION OF PACKAGED STARBURST REFLECTOR



FOCAL POINT AND THICKNESS PACKAGING CONSIDERTIONS

(3 Ring, 20 m D eff)



STARBURST COMMENTS

Low level of effort to date (Primarily a concept feasibility study)

Has potential for deploying 20 meter class reflectors from Shuttle-size cargo bay

Two basic deployment concepts

- o Synchronized mechanism
- o Distributed actuators

Further work needed

- o Detailed packaging study for both concepts
- o Deployment simulation for both concepts
- o Build demonstration model
- o Deployable support structure concept study
- o Dynamic & accuracy active control operation simulation studies

Center for Space Construction
Third Annual Symposium
November 21, 22, 1991

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