## An AsSESSMENT OF <br> Twilight Airglow Inversion Procedures Using Atmosphere Explorer Observations

NASA Grant NO. NAG 5-1502 163209
$p .81$

Principal Investigators*
I. C. McDade\# and W. E. Sharp

Space Physics Research Laboratory
Department of Atmospheric, Oceanic and Space Science
The University of Michigan

FINAL REPORT
April 1993

|  | N93-27483 |
| :--- | :--- |
|  |  |
|  |  |
| Uncles |  |
| G3/46 | 0163209 |

## TABLE OF CONTENTS

## Summary

1. Background to the Research Activities
2. The Twilight Airglow Inversion Algorithm
3. The VAE Observations and the Test Data Requirements

3
3.1 The Basic Data Requirements
3.2 The Visible Airglow Experiment $7320 \AA$ Observations
3.3 Primary Data Selection Criteria
4. Identification of Orbits Suitable for Testing the Inversion Algorithm 12
5. Inversion Code Modifications for Inverting the Satellite Data16
6. Results from the Inversions of Selected Satellite Orbits
6.1 Wide Channel Results - Orbit 6855
6.2 Narrow Channel Results - Orbit 7012
6.3 Results at Higher Levels of Solar Activity - Orbit 24564
7. Discussion and Conclusions
8. Acknowledgements
9. References

## FIGURES

Appendix 1. Table of AE-E Orbits with VAE $7320 \AA$ Observations
Appendix 2. Listing of program TWIVAEREAD
Appendix 3. Listing of program TWIFITTER

## SUMMARY

The aim of this research project was to test and truth some recently developed methods for recovering thermospheric oxygen atom densities and thermospheric temperatures from ground-based observations of the $7320 \AA \mathrm{O}^{+}(2 \mathrm{D}-2 \mathrm{P})$ twilight airglow emission. The research plan was to use twilight observations made by the Visible Airglow Experiment (VAE) on the Atmosphere Explorer ' E ' satellite as proxy groundbased twilight observations. These observations were to be processed using the twilight inversion procedures and the recovered oxygen atom densities and thermospheric temperatures were then to be examined to see how they compared with the densities and temperatures that were measured by the Open Source Mass Spectrometer and the Neutral Atmosphere Temperature Experiment on the satellite.

The activities performed under the one year performance period of the grant may be summarized as follows:
(1) A major survey of the Atmosphere Explorer ' $E$ ' data base was first performed in order to identify the orbits for which suitable Visible Airglow Experiment, Open Source Mass Spectrometer and Neutral Atmosphere Temperature Experiment observations existed.
(2) Satellite versions of the twilight airglow inversion program which allowed for the viewing geometry of the VAE observations were generated and tested using synthetic data. The inversion program was also modified to allow for the analysis of twilight observations which included contributions from regions of space with local solar zenith angles less than 90 degrees.
(3) The twilight observations made on selected orbits were inverted and the atomic oxygen densities and thermospheric temperatures recovered from these inversions were compared with the densities and temperatures measured at the satellite. The $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequencies, which are also recovered as part of the inversion process, were compared with the frequencies deduced using other methods.

The results of the study show that at both low and high levels of solar activity, the atomic oxygen densities and thermospheric temperatures recovered from the inversions are in reasonably good agreement with the in situ satellite data. The $\mathrm{O}^{+}(2 \mathrm{P})$ ionization
frequencies recovered for low levels of solar activity are in good agreement with previous evaluations, however, the ionization frequencies recovered from the twilight observations made closer to solar maximum exhibit a weaker than expected dependence on the solar $\mathrm{F}_{10.7}$ flux.

A full journal article describing the results of this work is in preparation and an abstract is being submitted to the American Geophysical Union for presentation at the Fall 1992 AGU Meeting .

## 1. BACKGROUND TO THE RESEARCH ACTIVITIES

Atomic oxygen is undoubtedly the most important neutral constituent of the thermosphere and during the last two decades satellite-borne mass spectrometer measurements have provided a great deal of information about the solar cycle, seasonal and diurnal variations of the thermospheric atomic oxygen densities. Unfortunately, however, many aspects of both the long term and short term variations, such as those caused by geomagnetic storms, are still not fully understood. At present there are no satellites in orbit providing atomic oxygen data and there is, therefore, a well recognized need to establish alternative methods for monitoring the thermospheric oxygen atom densities. This need has stimulated a renaissance of interest in twilight airglow studies and recent research has demonstrated that ground-based twilight observations of selected airglow emission features may provide a great deal of information about the long term and short term variations in thermospheric temperatures and thermospheric oxygen atom densities.

One of the most promising ground-based thermospheric monitoring techniques proposed during the last few years is based upon twilight observations of the $\mathrm{O}^{+}(2 \mathrm{D}-2 \mathrm{P})$ airglow emission at $7320 \AA$ (Fennelly et al., 1991; McDade et al., 1991). The method, originally pioneered by Meriwether et al. (1978) and Noxon and Norton (1979), would use twilight airglow emission rate measurements made at low elevation in the direction of the rising or setting sun to determine the oxygen atom densities and thermospheric temperatures. Unfortunately, as originally formulated, this method suffers from the limitation that detailed information about the solar EUV flux and the $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequencies at the time of the observations is required in order to recover the densities and temperatures. However, McDade et al. (1991) have demonstrated that this limitation may be overcome, and valuable information about the $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequencies can also be obtained, if the $7320 \AA$ twilight observations are made in two different viewing directions - one at low elevation towards the sun and the other at higher elevation ideally towards the local zenith. Preliminary ground-based measurements using this approach are now underway, however, there are at present no satellite measurements of the oxygen atom densities being made and it will not be possible, therefore, to directly verify or truth the atomic oxygen densities and thermospheric temperatures recovered using this technique. Fortunately, during the Atmosphere Explorer ' E ' mission simultaneous measurements of the $7320 \AA$ twilight airglow emission (Visible Airglow Experiment) were made together with measurements of the thermospheric oxygen atom densities
(Open Source Mass Spectrometer) and the thermospheric temperatures (Neutral Atmosphere Temperature Experiment). Since some of the Visible Airglow Experiment (VAE) observations were made in a multi-directional spin-scan mode they should closely resemble ground-based observations and may, therefore, be used to truth and test the $\mathrm{O}^{+}\left({ }^{2} \mathrm{D}-2 \mathrm{P}\right) 7320 \AA$ twilight inversion procedures. This report describes the results of a study carried out to assess the performance of the twilight inversion procedures using the Atmosphere Explorer ' $E$ ' data base.

## 2. THE TWILIGHT AIRGLOW INVERSION ALGORITHM

The twilight inversion procedures assessed in this work are discussed in detail by McDade et al. (1991) and are only briefly described here. The inversion algorithm is based upon the relatively complete understanding of the $\mathrm{O}^{+}(2 \mathrm{P})$ photochemistry that has emerged from the AE-C and AE-D missions (Rusch et al. 1976, 1977).

The $\mathrm{O}^{+}(2 \mathrm{P})$ ion, responsible for the airglow emission at 7620 and $7330 \AA$, is primarily produced under twilight conditions as a result of direct photoionization excitation of atomic oxygen by solar EUV photons with wavelengths less than $\sim 666 \AA$,

$$
\begin{equation*}
\mathrm{O}+\mathrm{h} v(\lambda<666 \AA) \rightarrow \mathrm{O}^{+}(2 \mathrm{P})+\mathrm{e} \tag{1}
\end{equation*}
$$

The ion may also be produced as a result of photoelectron impact ionization excitation of atomic oxygen, but this source is thought to make only a $\sim 10 \%$ contribution under most twilight conditions (Torr et al. 1990). The $\mathrm{O}^{+}\left({ }^{2} \mathrm{P}\right)$ ions are lost through the radiative decay process

$$
\begin{equation*}
\mathrm{O}^{+}(2 \mathrm{P}) \rightarrow \mathrm{O}^{+}(2 \mathrm{D}, 4 \mathrm{~S})+\mathrm{h} v(7320-30 \AA, 2470 \AA) \tag{2}
\end{equation*}
$$

and are primarily quenched at thermospheric altitudes by atomic oxygen and molecular nitrogen,

$$
\begin{array}{r}
\mathrm{O}^{+}(2 \mathrm{P})+\mathrm{O} \rightarrow \mathrm{O}^{+}+\mathrm{O} \\
\mathrm{O}^{+}(2 \mathrm{P})+\mathrm{N}_{2} \rightarrow \mathrm{O}^{+}+\mathrm{N}_{2} \text { or } \mathrm{O}+\mathrm{N}_{2}^{+} \tag{4}
\end{array}
$$

Because of this relatively simply photochemistry, the twilight $7320-30 \AA$ volume emission rate at any point in space defined by the altitude, z , and local solar zenith angle, $\beta$, may be expressed as

$$
\begin{equation*}
\mathrm{V}(\mathrm{z}, \beta)=\{\gamma \times \mathrm{A} \times \mathrm{P}(\mathrm{z}, \beta)\} /\left\{\mathrm{A}+\mathrm{k}_{0}[\mathrm{O}]_{\mathrm{z}}+\mathrm{k}_{\mathrm{N} 2}\left[\mathrm{~N}_{2}\right]_{\mathrm{z}}\right\} \tag{5}
\end{equation*}
$$

where $\mathrm{k}_{\mathrm{O}}$ and $\mathrm{k}_{\mathrm{N} 2}$ are the rate coefficients for quenching of $\mathrm{O}^{+}(2 \mathrm{P})$ by atomic oxygen and molecular nitrogen; A is the inverse radiative lifetime of the ${ }^{2} \mathrm{P}$ state; $\gamma$ is the branching ratio for emission of the $\left(2 \mathrm{D}_{5 / 2} \leftarrow 2 \mathrm{P}_{3 / 2,1 /}\right)$ and $\left(2 \mathrm{D}_{3 / 2} \leftarrow 2 \mathrm{P}_{3 / 2,1 / 2}\right)$ pair of doublets at $7320 \AA$ and $7330 \AA$; and $\mathrm{P}(\mathrm{z}, \beta)$ is the altitude and solar zenith angle dependent local $\mathrm{O}^{+}(2 \mathrm{P})$ volume production rate.

The $\mathrm{O}^{+}(2 \mathrm{P})$ volume production rate due to ionization excitation by EUV photons in a narrow wavelength interval centered on the wavelength $\lambda_{i}$ is given by

$$
\begin{equation*}
\mathrm{Q}_{\mathrm{i}}(\mathrm{z}, \beta)=[O]_{\mathrm{z}} \times \mathrm{f}_{\mathrm{i}} \times \mathrm{I}^{*} \times \exp \left[-\tau_{\mathrm{i}}(\mathrm{z}, \beta)\right] \tag{6}
\end{equation*}
$$

where $\tau_{i}$ is the optical depth for the radiation at wavelength $\lambda_{i}$ and $f_{i}$ is the fractional contribution made by radiation in this interval to the total $\mathrm{O}^{+}\left({ }^{2} \mathrm{P}\right)$ ionization frequency, $I^{*}$, at zero optical depth. The total $\mathrm{O}^{+}\left({ }^{2} \mathrm{P}\right)$ volume production rate, $\mathrm{P}(\mathrm{z}, \beta)$, may be obtained by summing the $\mathrm{Q}_{\mathrm{i}}(\mathrm{z}, \mathrm{\beta})$ over all wavelength intervals that contribute significantly to I*. This is achieved by binning the solar EUV spectrum into the wavelength intervals described by Torr et al., (1979). The contribution that photons in a given wavelength interval make to the total ionization frequency is calculated using only the spectral shape of a reference solar EUV spectrum.

As most of the twilight 7320-30 $\AA$ emission originates from altitudes above $\sim 250 \mathrm{~km}$ (Torr et al., 1990; Fennelly et al., 1991) the atomic oxygen densities may be approximated by a single exponential profile. The oxygen atom density at any altitude can then be expressed in terms of two parameters - the atomic oxygen scale height $\mathrm{H}_{0}$, which is determined by the thermospheric temperature, and the absolute oxygen atom number density, $\left[\mathrm{O}_{250}\right.$, at an arbitrary reference altitude of 250 km . Similarly, the molecular nitrogen density at each altitude may be expressed in terms of the density at a reference altitude of 250 km and the nitrogen scale height $\mathrm{H}_{\mathrm{N} 2}=(16 / 28) \times \mathrm{H}_{\mathrm{O}}$.

In the thermosphere most of the attenuation of the EUV flux is due to absorption by atomic oxygen and molecular nitrogen and for exponential O and $\mathrm{N}_{2}$ profiles, the optical depth at wavelength $\lambda_{i}$ may be obtained from the expression

$$
\begin{equation*}
\tau_{i}(\mathrm{z}, \beta)=\left\{[\mathrm{O}]_{\mathrm{a}} \times \mathrm{H}_{0} \times \sigma_{\mathrm{i}}^{0} \times \operatorname{Ch}\left(\beta, \mathrm{H}_{0}\right)\right\}+\left\{\left[\mathrm{N}_{2}\right]_{\mathrm{a}} \times \mathrm{H}_{\mathrm{N} 2} \times \sigma_{\mathrm{i}}^{\mathrm{N} 2} \times \operatorname{Ch}\left(\beta, \mathrm{H}_{\mathrm{N} 2}\right)\right\} \tag{7}
\end{equation*}
$$

where $\sigma_{i}{ }^{\circ}$ and $\sigma_{i}{ }^{N_{2}}$ are the total O and $\mathrm{N}_{2}$ cross sections at $\lambda_{\mathrm{i}} ; \operatorname{Ch}(\beta, \mathrm{H})$ is the grazing incidence Chapman function and $\left[\mathrm{O}_{\mathrm{a}}\right.$ and $\left[\mathrm{N}_{2}\right]_{\mathrm{a}}$ are the O and $\mathrm{N}_{2}$ number densities at the minimum ray height of the grazing solar radiation.

By integrating the volume emission rates given by equation 5 along the line-of-sight corresponding to a particular ground-based or satellite-borne twilight observation, the
measured 7320-30 $\AA$ column emission rate may be expressed in terms of the well known physical quantities appearing in (5), (6) and (7) and the four important, and variable, geophysical parameters $I^{*}, \mathrm{H}_{\mathrm{O}},\left[\mathrm{O}_{250}\right.$ and $\left[\mathrm{N}_{2}\right]_{250}$. Consequently, it is possible in principle to deduce the later four quantities from a series of twilight observations by finding the set of four parameters that best reproduces the observations. In practice, however, the two parameters $I^{*}$ and $\left[\mathrm{O}_{250}\right.$ are strongly coupled and it is really only possible to separate these two quantities if the observational data consists of a series of measurements made over a range of solar depression angles at two different elevation angles, ideally one at low elevation towards the azimuth of the sun and the other towards the zenith (McDade et al., 1991). It also turns out that the twilight emission rates are not particularly sensitive to the molecular nitrogen densities and the problem may be reduced to one of finding $\mathrm{I}^{*}, \mathrm{H}_{0}$, and $\left[\mathrm{O}_{250}\right.$ if independently measured nitrogen densities or nitrogen densities from a standard atmospheric model are substituted for $\left[\mathrm{N}_{2}\right]_{250}$.

The problem of finding the values for the parameters $I^{*}, H_{0}$, and $[\mathrm{O}]_{250}$ that best reproduce a given set of twilight observations may be solved using a standard non-linear least squares fitting procedure such as the Marquardt gradient-expansion method described by Bevington (1969) and Press et al. (1986). This iterative procedure efficiently searches for the set of parameters that optimizes the agreement between a model and a set of observations through minimization of the $\chi 2$ merit function.

## 3. THE VAE OBSERVATIONS AND THE TEST DATA REQUIREMENTS

### 3.1 The Basic Data Requirements

In order to make a meaningful assessment of the inversion procedures described in Section 2 it was important to use satellite observations that were obtained with viewing geometries that were as similar as possible to those that would be used to make the ground-based twilight measurements. Ideally, the ground-based observations would consist of a series of twilight brightness measurements made over a time interval during which the solar depression angle at the ground varied between 5 and 20 degrees (McDade et al. , 1991). Approximately one half of these observations would be made at an elevation angle of $\sim 20^{\circ}$ towards the azimuth of the setting, or rising, sun and the other half would be made towards the local zenith. The idealized ground-based observing geometry for which similar satellite observations were to be found is illustrated in Figure 1 a .

### 3.2 The Visible Airglow Experiment 7320-30 A Observations

The Visible Airglow Experiment (VAE) on the Atmosphere Explorer 'E' satellite (AEE) was a filter wheel airglow photometer designed to measure various thermospheric emission features during both daytime and nighttime conditions. The photometer had two distinct optical channels, a high sensitivity channel with a large field of view ( $3^{\circ}$ half cone angle) and a low sensitivity channel with a narrow field of view ( $3 / 4^{\circ}$ half cone angle). The fields of view of the two channels were oriented at 90 degrees from each other. The counts from the narrow channel (channel 1) were integrated over a period of 32 msec and those from the wide channel (channel 2) were integrated over a 125 msec interval. The instrument operated in a number of different modes depending on the filter wheel position which could be held fixed to continuously monitor a particular airglow feature or stepped through a number of different interference filters at various stepping rates. Continuous observations of the airglow emission near $7320 \AA$ were made in two of the eight possible VAE fixed wheel modes. In one of these modes, known as mode '73F6', the $7320 \AA$ observations were made with the narrow channel of the instrument (channel 1); in another fixed wheel mode, mode ' 55 F 7 ', the wide channel (channel 2) was used to make the $7320 \AA$ observations.

During normal satellite operation the VAE instrument was oriented so that the narrow channel pointed aft of the spacecraft and the wide channel pointed towards the earth.

However, for a significant part of the mission the spacecraft was operated in 'skid' or 'cartwheel' spin modes. In the 'skid' mode the satellite spin angular momentum vector was anti-parallel to the orbital angular momentum vector; this is referred to as the 'normal' spin mode. In the 'cartwheel' mode the satellite spin angular momentum vector was parallel to the orbital angular momentum vector and this is referred to as the 'inverted' spin mode. In either spin mode the two VAE channels scanned through all zenith angles within the orbital plane. Consequently, when the satellite passed through the terminator in the spinning mode with an active $7320 \AA$ channel the instrument made twilight observation of the $\mathrm{O}^{+}(2 \mathrm{D}-2 \mathrm{P})$ airglow in a manner similar to a ground-based twilight monitoring station as shown in Figure 1 b .

### 3.3 Primary Data Selection Criteria

Given the nature of the basic data requirements and the operational characteristics of the AE-E VAE instrument the first task to be performed was to identify the satellite orbits which satisfied the following primary criteria:
(1) The satellite had to be operating in a spinning mode in a low altitude, near circular, orbit.
(2) The VAE instrument had to be operational as the satellite passed through the dusk or dawn terminator and the instrument had to be observing the $7320 \AA$ airglow in either Channel 1 or Channel 2 .
(3) The Open Source Mass Spectrometer and the Neutral Atmosphere Temperature Experiment both had to be operating and providing good atmospheric density and temperature data.

## 4. IDENTIFICATION OF ORBITS SUITABLE FOR TESTING THE INVERSION ALGORITHM

The first step in the search for suitable orbits for testing the inversion algorithm was to find the AE-E orbits during which the VAE instrument was observing the $7320 \AA$ airglow and the satellite was operating in the spin mode. These orbits are listed in the table of Appendix 1 along with some pertinent information identifying the VAE channel that was observing at $7320 \AA$ - either 1 or 2 ; the spin mode of the satellite - either normal or inverted; the on/off times of the instrument and the satellite altitude and local times associated with the instrument on/off times. Having identified these orbits the next step was to check that the VAE instrument was operating when the satellite passed through the terminator and to confirm that density and temperature data from the Open Source Mass Spectrometer (OSS) and the Neutral Atmosphere Temperature Experiment (NATE) were available at that time. This was achieved using the AE database software available at the University of Michigan Space Physics Research Laboratory and the Atmosphere Explorer United Abstract Data files. The AE data base program 'NEWLIST' was used to examine the VAE, OSS and NATE data on the United Abstract (UA) files for all of the orbits listed in Table A1. The UA data files contain a summary of the data from all the AE instruments with a data point for every 15 second interval. The program NEWLIST allows the data satisfying specific selection criteria to be extracted to a file or plotted on a visual display unit. It should be noted that when the satellite was operating in a spin mode the UA files contain only the VAE observations made when the instrument channels were pointing towards the zenith. By running the NEWLIST program and selecting only the data acquired when the solar zenith angle at the satellite was between 85 and 145 degrees the AE-E orbits satisfying the criteria described above were identified. These orbits are listed in Table 1 which gives the date and orbit number as well as the number of the channel that was observing at $7320 \AA$. Table 1 also lists the Universal Time in seconds for the start and end of each twilight observing sequence and an indication of whether the satellite was passing from the dayside to the nightside (i.e., sunset) or from the nightside to the dayside (sunrise).

Once the twilight passes listed in Table 1 were identified the actual VAE observations had to be examined in detail to make sure that the instrument was operating normally and to ascertain if the observations were of sufficient quality to satisfy our needs. Since the UA files only contained summary data of the VAE brightness measurements in the zenith, the full time resolution VAE data files had to be inspected.

This was performed using the Fortran program 'VAEREAD' which unpacks the VAE data files and extracts the VAE brightness measurements and the ancillary instrument and orbital data. A listing of the version of VAEREAD used in this project, TWIVAEREAD, is attached as Appendix 2. The program reads the channel 1 or channel 2 photometer counts, converts the observed counts into Rayleighs and makes zodiacal and galactic background corrections where possible. The program creates two output text (ASCI) files. One of the files lists sequentially the following quantities:
(1) The universal time in milliseconds
(2) The observed $7320 \AA$ brightness in captured Rayleigh units
(3) The estimated error in the observed brightness based on the photometer count rate
(4) The satellite altitude in kilometers
(5) The zenith angle of the photometer line of sight in degrees
(6) The solar zenith angle at the satellite in degrees

The second output file contains the following quantities:
(1) The universal time in milliseconds
(2) The $x, y$ and $z$ coordinates of the satellite position in the Geocentric Equatorial Inertial (GEI) system (see Russell, 1971)
(3) The $x, y$ and $z$ coordinates of the sun in the GEI system
(4) The $x, y$ and $z$ coordinates of the photometer line-of-sight in the GEI system

Detailed examination of the high resolution VAE data for the orbits listed in Table 1 revealed that in many instances, particularly in the case of the channel 1 data, the observations were seriously and irrevocably contaminated by stars or that the zodiacal and galactic background corrections could not be performed. Furthermore, although the VAE instrument was equipped with a sophisticated baffle system, close examination of the data revealed that the twilight observations made at low elevations towards the sun were often contaminated by scattered sunlight. Because of these various undesirable effects most of the twilight passes listed in Table 1 had to be rejected. Having identified the passes which contained potentially useful data a final selection criterion then had to applied.

As described in Section 3 the ground-based thermospheric monitoring technique, and the twilight inversion procedure, require twilight $7320 \AA$ measurements made in the zenith and at lower elevation towards the azimuth of the sun, i.e. the observations are made in a plane that is perpendicular or nearly perpendicular to the plane of the terminator. However, because of seasonal effects and the 20 degree inclination of the AE-E orbit, the angle between the orbital plane of the satellite and the plane of the terminator varied considerably and on many of the twilight passes the plane of the spin scan observations was not perpendicular to, or nearly perpendicular to, the plane of the terminator. Consequently, only a small number of the twilight passes listed in Table 1 actually provided observations of sufficient quality made under the appropriate geometry for testing the inversion algorithm. Of these the potentially most useful observations were the channel 2 sunset observations made on orbit 6855 and the channel 1 sunset observations made on orbit 7012.

Table 1.
AE-E spin twilight passes with OSS, NATE and VAE $7320 \AA$ observations

| Date yyddd | Orbit <br> \# | Channel \# | Spin mode | Start time | End time | Condition |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 77003 | 5811 | 2 | normal | 25725 | 27165 | sunrise |
| 77003 | 5874 | 1 | normal | 19005 | 19830 | sunrise |
| 77007 | 5878 | 2 | normal | 40500 | 41295 | sunrise |
| 77007 | 5884 | 1 | normal | 72585 | 73485 | sunrise |
| 77009 | 5912 | 2 | normal | 50085 | 50910 | sunrise |
| 77009 | 5913 | 2 | normal | 53430 | 54540 | sunset |
| 77009 | 5913 | 2 | normal | 55155 | 18915 | sunset |
| 77016 | 6019 | 2 | normal | 18495 | 51165 | sunset |
| 77016 | 6025 | 1 | normal | 50565 | 72660 | sunset |
| 77016 | 6029 | 2 | normal | 71955 | 69495 | sunrise |
| 77040 | 6415 | 2 | normal | 68505 | 53865 | sunset |
| 77047 | 6525 | 2 | normal | 52785 | 54975 | sunrise |
| 77047 | 6525 | 2 | normal | 54540 | 9345 | sunrise |
| 77052 | 6597 | 1 | normal | 8910 | 84315 | sunrise |
| 77052 | 6612 | 2 | normal | 83970 | 20505 | sunset |
| 77063 | 6776 | 1 | inverted | 19590 | 22410 | sunrise |
| 77063 | 6777 | 1 | normal | 21495 | 13560 | sunset |
| 77068 | 6855 | 2 | inverted | 12660 | 74235 | sunset |
| 77077 | 7012 | 1 | normal | 73320 | 76140 | sunrise |
| 77077 | 7012 | 1 | normal | 75225 | 66810 | sunset |
| 77081 | 7075 | 2 | normal | 65850 | 68685 | sunrise |
| 77081 | 7075 | 2 | normal | 67725 | 48870 | sunset |
| 77087 | 7168 | 1 | normal | 47970 | 52320 | sunrise |
| 77087 | 7169 | 1 | normal | 51405 | 22455 | sunset |
| 77100 | 7372 | 2 | inverted | 21795 | 27165 | sunset |
| 77100 | 7373 | 2 | inverted | 26790 | 4065 | sunset |
| 77108 | 7497 | 1 | inverted | 3045 | 5895 | sunrise |
| 77108 | 7497 | 1 | inverted | 4875 | 61005 | sunset |
| 77132 | 7895 | 1 | inverted | 60435 | 62895 | sunrise |
| 77132 | 7895 | 1 | inverted | 61935 | 65820 | sunset |
| 77132 | 7896 | 1 | inverted | 945 | 1305 | sunset |
| 77138 | 7980 | 2 | inverted | 1995 | 3075 | sunrise |
| 77138 | 7980 | 2 | inverted | 59115 | 60545 | sunset |
| 77142 | 8055 | 1 | inverted | 60545 | 61980 | sunrise |
| 77142 | 8055 | 1 | inverted | 76115 | 77550 | sunrise |
| 77146 | 8122 | 2 | inverted | 24075 | 24990 | sunset |
| 77155 | 8257 | 1 | inverted | 50265 | 50565 | sunset |
| 77176 | 8600 | 1 | inverted | 51435 | 52440 | sunrise |
| 77176 | 8600 | 1 | inverted | 54975 | 55645 | sunset |
| 77176 | 8601 | 1 | inverted | 6960 | 8375 | sunset |
| 77181 | 8672 | 2 | inverted | 45045 | 45525 | sunrise |
| 77181 | 8672 | 2 | inverted | 55620 | 56820 | sunrise |
| 77364 | 11617 | 2 | normal | 59355 | 60450 | sunset |
| 77364 | 11618 | 2 | normal | 59355 |  |  |

## 5. INVERSION CODE MODIFICATIONS FOR INVERTING the Satellite Data

As already mentioned, the ground-based twilight inversion algorithm is designed to deal with observations made in the zenith and at low elevation in the azimuth of the sun. Since the spin plane of the twilight VAE observations was rarely oriented perpendicular to the plane of the terminator, the inversion algorithm had to be modified to allow for the finite angle between the azimuth of the sun and the photometer line-of-sight.

In order to allow for this effect the program which was used to read the VAE data files, TWIVAEREAD, was modified to extract the Geocentric Equatorial Inertial coordinates of (i) the satellite position vector, (ii) the sun vector, and (iii) the vector of the VAE photometer line-of-sight, as well as the observed $7320 \AA$ column emission rates and photometer zenith angles. These vectors were then used within the inversion algorithm to calculate the local solar zenith angle at various intervals along the photometer line-of-sight. The section of code dealing with this problem is incorporated in the line integral calculation performed by the procedure 'BRIGHT which appears on pages 5 and 6 of the listing given in Appendix 3.

Other modifications had to be made to the inversion code to allow for the fact that many of the VAE observations were made at smaller solar depression angles than would be accessible from the ground. In the case of ground-based observations tropospheric scattering of sunlight makes it very difficult to obtain good $\mathrm{O}^{+}$(2D-2P) 7320-30 $\AA$ measurements until the solar depression angle at the observing site is greater than about 7 degrees for zenith observations and about 12 degrees for low elevation angle sunward observations (Meriwether et al. ,1978; Noxon and Norton, 1979; Fennelly et al., 1991; McDade et al., 1991). However, tropospheric scattering does not interfere with the satellite measurements and good twilight data were obtained for satellite solar depression angles down to zero degrees. It was considered valuable to include these 'early' twilight observations but the early observations made at low photometer elevation angles inevitably included contributions from regions of space lying on the dayside of the terminator where the local solar zenith angle, $\beta$, was less than 90 degrees. The line integral calculation section of the original inversion code, procedure BRIGHT, was therefore modified to deal with this situation by including brightness contributions from both sides of the terminator. For contributing elements with local solar zenith angles greater than 90 degrees the grazing incidence Chapman function was used to calculate the optical depth as explained in Section 2 (equation 7). For elements with local solar
zenith angles less than 90 degrees the normal Chapman function was used and the optical depth at each wavelength, $\lambda_{i}$, was calculated from the expression

$$
\begin{equation*}
\tau_{i}(\mathrm{z}, \beta)=\left\{[\mathrm{O}]_{\mathrm{z}} \times \mathrm{H}_{0} \times \sigma_{\mathrm{i}}^{\mathrm{o}} \times \operatorname{Ch}\left(\beta, \mathrm{H}_{0}\right)\right\}+\left\{\left[\mathrm{N}_{2}\right]_{\mathrm{z}} \times \mathrm{H}_{\mathrm{N} 2} \times \sigma_{\mathrm{i}}^{\mathrm{N} 2} \times \operatorname{Ch}\left(\beta, \mathrm{H}_{\mathrm{N} 2}\right)\right\} \tag{8}
\end{equation*}
$$

where $z$ is the altitude of the contributing element and $[\mathrm{O}]_{z}$ and $\left[\mathrm{N}_{2}\right]_{z}$ are the inferred atomic oxygen and molecular nitrogen densities at that height. The relevant section of the inversion code appears on pages 5 and 6 of Appendix 3.

## 6. RESULTS FROM THE INVERSIONS OF SELECTED SATELLITE ORBITS

In an ideal world it would be more desirable to test the inversion algorithm using the observations made with the narrow channel of the VAE instrument (channel 1). However, the narrow channel was about 50 times less sensitive than the wide channel and the advantages of using the channel with the smaller field of view (and shorter integration period) were strongly outweighed by the much higher signal to noise ratio in the channel 2 observations. Nevertheless, it was considered instructive to test the algorithm using both the wide channel and the narrow channel observations.

### 6.1 Wide Channel Results - Orbit 6855

For the purposes of this study the best VAE channel 2 twilight observations were the sunset observations made on orbit 6855 on day 68 of 1977. When the measurements were made the satellite was in a near circular orbit at a latitude of $12{ }^{\circ} \mathrm{N}$, a longitude of $215^{\circ} \mathrm{E}$ and an altitude of 257 km . The solar $\mathrm{F}_{10.7}$ flux value for the day was 80.3 and the Ap index was 38 . The angle between the orbital plane and the terminator was approximately 80 degrees as the satellite crossed the terminator and the angle between the azimuth of the photometer scan and the azimuth of the sun was 10.2 degrees. Although it was important to have the latter angle as small as possible (see Section 4), it did mean that the observations made at the lowest elevations in the sunward direction were prone to contamination by scattered sunlight. To avoid the danger of using observations with solar contamination we only considered the observations which were made when the angle between the sun and photometer line-of-sight was greater than 40 degrees. As a result, the analysis was restricted to zenith observations and sunward observations made at elevation angles greater than $\sim 35$ degrees. In Figure 2 we show part of the sequence of $\mathrm{O}^{+}(2 \mathrm{D}-2 \mathrm{P}) 7320 \AA$ column emission rates measured during the sunset pass of orbit 6855 . The solid symbols show the $7320 \AA$ emission rates measured when the photometer elevation angle at the midpoint of the sample integration period was within $\pm 5$ degrees of the zenith direction; the open symbols show the emission rates measured in the sunward direction when the photometer elevation angle was between 35 and 45 degrees. Both sets of observations are plotted against the solar zenith angle at the satellite which was increasing with time. The satellite was spinning at approximately 4 revolutions per minute and the measurements shown in Figure 2 span a total observing period of about 5 minutes.

A comparison of the data shown in Figures 2 and 3 reveals the similarity between the VAE twilight measurements and the simulated ground-based observations discussed by McDade et al., (1991). It should be noted, however, that relative to the zenith observations the sunward observations of Figure 2 are weaker than those of Figure 3. This is primarily due to the fact that the sunward VAE observations were made at an elevation angle of $40 \pm 5$ degrees and the sunward simulations were calculated for an elevation of only 20 degrees. It is also evident that the uncertainties in the VAE measurements are considerably larger than those of the ground-based simulations in spite of the fact that the satellite observations were made in the absence of any tropospheric background scattering. However, it has to be recognized that each of the satellite observations was obtained with a sample integration period of only 0.125 seconds whereas the ground-based simulations were calculated for a one minute integration period using an instrument with a larger throughput. It should also be pointed out that not all of the scatter evident in the VAE observations is to be associated with noise because much of it is due to variations in the photometer elevation angle within the plotted $\pm 5$ degree elevation angle bands. This occurs because the satellite was spinning at approximately 24 degrees per second and the photometer counts were sampled every 0.125 seconds, therefore, there were usually three observations made on each spin between the elevation angles of 35 and 45 degrees and three observations made within $\pm 5$ degrees of the zenith. Since the observed twilight brightness depends on both the photometer elevation angle and the solar depression angle, the $\pm 5$ degree spread in elevation angles contributes to the scatter which should be greater in the case of the sunward viewing observations.

When the $7320 \AA$ emission rates of Figure 2 were corrected for the $\mathrm{O}^{+}\left({ }^{2} \mathrm{D}-2 \mathrm{P}\right) 7320$ $\AA$ and $7330 \AA$ doublet filter capture functions and processed using the twilight inversion program (see Appendix 3) the algorithm returned the following set of fitting parameters :
(i) an atomic oxygen scale height of $\mathrm{H}_{0}=62 \pm 3.4 \mathrm{~km}$
(ii) an oxygen atom density at 250 km of $[\mathrm{O}]_{250}=1.4 \pm 0.3 \times 10^{9} \mathrm{~cm}^{-3}$
and (iii) an unattenuated $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequency of $\mathrm{I}^{*}=6.2 \pm 0.6 \times 10^{-8} \mathrm{sec}^{-1}$
The fitting parameters were not found to depend significantly upon the first guess values used to initiate the inversion procedure although the number of iterations required to reach convergence did of course depend upon the initial guess. The fits to
the zenith and sunward observations obtained using these best fit parameters are shown in Figures 4 and 5. In the case of the sunward observations, Figure 5, the fit is 'saw toothed' rather than smooth because of the $\pm 5$ degree spread in the elevation angles discussed above. Since the twilight brightness should not vary so strongly between +5 and -5 degrees of the zenith this effect does not show up in the fit to the zenith observations.

Clearly, the parameters recovered from the fit to the orbit 6855 observations do reproduce fairly well the input data. More significantly, however, they also reproduce the orbit 6855 observations that were not used in the inversion, i.e. the recovered fitting parameters also reproduce the VAE observations that were not used to obtain the parameters. This is illustrated in Figures 6,7,8 and 9 which show the fits to the sunward observations that were made at elevation angles in the ranges 45 to 55 degrees, 55 to 65 degrees, 65 to 75 degrees and 75 to 85 degrees.

The neutral atmospheric temperatures measured at 257 km by the Neutral Atmosphere Temperature Experiment (NATE) as the satellite passed through the sunset terminator on orbit 6855 varied between about 870 K and 950 K . Since the recovered atomic oxygen scale height of $62 \pm 3.4 \mathrm{~km}$ is equivalent to an exospheric temperature of $970 \pm 50 \mathrm{~K}$ we see that the thermospheric temperature inferred from the twilight observations is in very good agreement with the NATE temperature measurements. Similarly, the average atomic oxygen density measured by the Open Source Mass Spectrometer (OSS) during the twilight pass was $9 \times 10^{8} \mathrm{~cm}^{-3}$ at 257 km which compares very favorable with the inferred density of $1.2 \pm 0.3 \times 10^{9} \mathrm{~cm}^{-3}$ based on the recovered density at 250 km and a scale height of 62 km . In Figure 10 we show how the atomic oxygen density profile constructed from the recovered fitting parameters $\mathrm{H}_{\mathrm{O}}$ and $\left[\mathrm{O}_{250}\right.$ compares with the profile based on the NATE temperature and the OSS density measurements. Figure 10 also shows the atomic oxygen density profile given by the MSIS-86 model (Hedin, 1987) for the sunset conditions on orbit 6855.

The value for the unattenuated $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequency obtained from the inversion of the orbit 6855 channel 2 observations will be discussed in Section 7.

### 6.2 Narrow Channel Results - Orbit 7012

The best channel 1 twilight $7320 \AA$ observations were those obtained on the sunset pass of orbit 7012 on day number 77 of 1977 . When these observations were made the satellite was in a circular orbit at a latitude of $16^{\circ} \mathrm{N}$, a longitude of $322^{\circ} \mathrm{E}$ and an altitude of 251 km . The solar $\mathrm{F}_{10.7}$ flux value for the day was 75 , the Ap index was 11 and the angle between the azimuth of the photometer scan plane and the azimuth of the sun was 14.5 degrees. As already mentioned the signal to noise ratios of the channel 1 observations were very much lower than those of the channel 2 observations. This is clearly illustrated in Figure 11 which shows the orbit 7012 channel 1 observations made within $\pm 5$ degrees of the zenith direction. Because of the low signal to noise ratios in the data it was not possible to simply invert the channel 1 observations made within $\pm 5$ degrees of the zenith and between 35 and 45 degrees elevation towards the sun, i.e. the very large uncertainties associated with the recovered fitting parameters rendered the inversion meaningless. However, somewhat more meaningful results were obtained when all of the observations acquired between elevations of 35 degrees and the zenith were considered. The entire set of observations obtained between satellite solar zenith angles of 90 and 105 degrees in the zenith and sunward in the elevation angle bands 35 -$45,45-55,55-65,65-75$ and 75-85 degrees are shown in Figure 12.

When this entire set of observations was inverted the inversion algorithm returned the following set of best fit parameters:
(i) an atomic oxygen scale height of $\mathrm{H}_{0}=76 \pm 14 \mathrm{~km}$
(ii) an oxygen atom density at 250 km of $[\mathrm{O}]_{250}=7.0 \pm 0.4 \times 10^{8} \mathrm{~cm}^{-3}$
and (iii) an unattenuated $\mathrm{O}^{+}\left({ }^{(2 \mathrm{P}}\right)$ ionization frequency of $\mathrm{I}^{*}=9.6 \pm 3.5 \times 10^{-8} \mathrm{sec}^{-1}$
The fit to the entire set of orbit 7012 observations obtained using these parameters is shown by the solid line through the smoothed data points in Figure 12.

The average neutral atmospheric temperature measured at 251 km by the Neutral Atmosphere Temperature Experiment (NATE) as the satellite passed through the sunset terminator on orbit 7012 was 880 K and the average atomic oxygen density measured by the Open Source Mass Spectrometer (OSS) during the same period was $1.0 \times 10^{9} \mathrm{~cm}^{-3}$. Clearly, the recovered oxygen density of $7.0 \pm 0.4 \times 10^{8} \mathrm{~cm}^{-3}$ at 250 km agrees with the OSS measured density within the uncertainty limits. The recovered atomic oxygen scale
height of $76 \pm 14 \mathrm{~km}$ is equivalent to an exospheric temperature of $1180 \pm 220 \mathrm{~K}$ which is somewhat hotter than the NATE measured temperature of 880 K .

### 6.3 Results at Higher Levels of Solar Activity - Orbit 24564

In order to rigorously assess the twilight inversion procedures it was considered highly desirable to test the inversion algorithm using VAE observations made at both low and high levels of solar activity. Unfortunately, most of the twilight passes which satisfied the primary selection criteria outlined in Section 3, and which contained data of a sufficiently high quality, were made at low levels of activity. However, there were a small number of passes made in 1980 close to solar maximum for which oxygen atom data were not available but which did involve channel 2 observations made when the satellite was spinning. One of these high activity orbits which contained apparently good data was orbit number 24564 on day 100 of 1980 . The twilight observations made on the sunset pass of orbit 24564 towards the zenith and towards the sun at an elevation of $40 \pm 5$ degrees are shown in Figure 13. When these observations were made the AE-E satellite was in a circular orbit at a latitude of $6.6^{\circ} \mathrm{N}$, a longitude of $80.4^{\circ} \mathrm{E}$ and an altitude of 419 km . The solar $\mathrm{F}_{10.7}$ flux value for the day was 244 , the Ap index was 20 , the angle between the azimuth of the photometer scan plane and the azimuth of the sun was 3.5 degrees and the satellite was operating in the inverted spin mode.

When the orbit 24564 column emission rates shown in Figure 13 were inverted the inversion procedure returned the following set of best fit parameters:
(i) an atomic oxygen scale height of $\mathrm{H}_{0}=94 \pm 8 \mathrm{~km}$
(ii) an oxygen atom density at 250 km of $[\mathrm{O}]_{250}=2.5 \pm 0.3 \times 10^{9} \mathrm{~cm}^{-3}$
and (iii) an unattenuated $\mathrm{O}^{+(2 \mathrm{P}}$ ) ionization frequency of $\mathrm{I}^{*}=9.3 \pm 1.0 \times 10^{-8} \mathrm{sec}^{-1}$
For the inversion of this data the shape of the EUV flux spectrum used in the algorithm (see Appendix page 14) was based on the 79050 spectrum reported by Torr et al. (1979), however, the results obtained from inversions using the shape of the standard F74113 spectrum of Hinteregger (1977) were not significantly different. The fits to the zenith and sunward observations obtained using the best fit parameters listed above are shown in Figures 14 and 15.

The average neutral atmospheric temperature measured by the Neutral Atmosphere Temperature Experiment (NATE) as the satellite passed through the sunset terminator on orbit 24564 was 1630 K and this compares quite favorable with the temperature of $1470 \pm 120 \mathrm{~K}$ inferred from the recovered atomic oxygen scale height of $94 \pm 8 \mathrm{~km}$. Unfortunately, OSS oxygen atom data was not available for this orbit but the recovered density of $2.5 \pm 0.3 \times 10^{9} \mathrm{~cm}^{-3}$ at 250 km compares very well with the density of $2.2 \times$ $10^{9} \mathrm{~cm}^{-3}$ given for the conditions by the MSIS-86 model (Hedin, 1987). However, as we will discuss in the next section, the recovered $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequency of $\mathrm{I}^{*}=9.3 \pm$ $1.0 \times 10^{-8} \mathrm{sec}^{-1}$ is substantially lower than might be expected for conditions close to solar maximum.

## 7. DISCUSSION AND CONCLUSIONS

The results described in the previous section clearly demonstrate that the inversion procedures for recovering thermospheric temperatures and atomic oxygen densities from bi-directional ground-based measurements of the $\mathrm{O}^{+}\left({ }^{(2 \mathrm{D}-2 \mathrm{P})} 7320 \AA\right.$ twilight airglow emission performed well when tested with the proxy satellite data. The atomic oxygen densities recovered from the inversions are in reasonable good agreement with the densities measured by the Open Source Mass Spectrometer on the AE-E satellite. The thermospheric temperatures inferred from the recovered atomic oxygen scale heights are also in reasonably good agreement with the measurements made on the satellite by the Neutral Atmosphere Temperature Experiment. Furthermore, the temperatures are also in good agreement with the temperatures that have been deduced using other techniques. This is illustrated in Figure 16 which shows how the temperatures recovered for orbits 6855,7012 and 24564 compare with the temperatures deduced by Yee and Abreu (1982) from an analysis of the late twilight $7320 \AA$ zenith intensities measured on these and other AE-E orbits.

The unattenuated $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequencies recovered for orbits 6855 and 7012 are in good agreement with previous evaluations but the ionization frequency recovered for orbit 24564 is somewhat smaller than might be expected for conditions close to solar maximum. The solar cycle dependence of the $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequencies has been studied by Torr et al. (1979) who used the solar EUV flux measurements on the Atmosphere Explorer satellites (Hinteregger., 1977) to calculate the ionization frequencies on five selected days during the 1974 to 1979 period. Abreu et al. (1980) have also investigated the solar cycle dependence of the $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequencies and used dayglow $7320 \AA$ measurements made with the VAE instrument to determine the frequencies during the increasing phase of solar cycle 21 . The ionization frequencies obtained from the work of Torr et al. (1979) and Abreu et al. (1980) are shown in Figures 17 and 18 where they are compared with the frequencies recovered here from the VAE twilight observations on orbits 6855, 7012 and 24564. Clearly, the frequencies recovered from orbits 6855 and 7012 are consistent with what should be expected at low levels of activity but the orbit 24564 frequency is not in keeping with the trends in the previous evaluations. Because of the lack of appropriately conditioned high activity twilight observations it is difficult to determine whether or not the seemingly low ionization frequency for orbit 24564 is indicative of a problem with the inversion algorithm. It is important to note, however, that the atomic oxygen scale height and
densities recovered from the inversion of the orbit 24564 data are in good agreement with the temperatures measured on the satellite and the densities predicted by the MSIS86 model. We should also point out that if the orbit 24565 observations are inverted with the $\mathrm{O}^{+}\left({ }^{2} \mathrm{P}\right)$ ionization frequency constrained to a value that is in keeping with the trends shown in Figure 18, then the recovered atomic oxygen scale height and densities are no longer in good agreement with the measured and modelled densities and temperatures. For example, if the ionization frequency is constrained to $1.3 \times 10^{-7} \mathrm{sec}^{-1}$ then the inversion algorithm returns an oxygen atom density at 250 km of $2.5 \times 10^{9} \mathrm{~cm}^{-3}$ and a thermospheric temperature of only 1320 K . It is possible, however, that the seemingly low value for the orbit 24564 ionization frequency is simply a reflection of (a) the natural variability of this quantity and (b) an incomplete correlation between the $\mathrm{O}^{+}\left({ }^{2} \mathrm{P}\right)$ ionization frequencies and the $\mathrm{F}_{10.7}$ radio flux. We do note, for example, that the relative displacement of the orbit 24564 frequency from the trend line in Figure 18 is not inconsistent with the scatter in the measured frequencies at low $\mathrm{F}_{10.7}$ flux values.

## 8. ACKNOWLEDGEMENTS

Support for this work under NASA Grant No. NAG 5-1502 to the University of Michigan's Space Physics Research Laboratory is gratefully acknowledged. We would also like to thank the Principal Investigators and Co-Investigators associated with the Visible Airglow Experiment (PIs - P. B. Hays and V. J. Abreu), the Neutral Atmosphere Temperature Experiment (PI - N. W. Spencer) and the Open Source Mass Spectrometer (PI - A. O. Neir) for providing an excellent aeronomical data base and for making the results of their experiments freely available for this work. We would also like to acknowledge the valuable contributions made to the study by Kris Kontz who participated in the project during the summer of 1991 as part of the National Science Foundation's Research Experience for Undergraduates Program at the University of Michigan. We are also greatly indebted to Edward Hume and Gerry Schmitt of the Space Physics Research Laboratory for their help with the AE and VAE data base software. Finally, very special thanks are due to Sam Yee for the excellence of his advice, his encouragement and his interest in the project at all times.

## 9. REFERENCES

Abreu, V. J., W. R. Skinner, and P. B. Hays, Airglow measurements of the variation of the $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequency during solar cycle 21 , Geophys. Res. Lett., 7, 109, 1980.

Bevington, P. R., Data Reduction and Analysis for the Physical Sciences, McGraw-Hill Book Company, New York, 1969.

Fennelly, J. A., D. G. Torr, P. G. Richards, M. R. Torr, and W. E. Sharp, A method for the retrieval of atomic oxygen density and temperature profiles from ground based measurements of the $\mathrm{O}^{+}(2 \mathrm{D}-2 \mathrm{P}) 7320-\AA$ twilight airglow, J. Geophys. Res., 96, 1263, 1991.

Hedin, A. E., MSIS-86 thermospheric model, J. Geophys. Res., 92, 4649, 1987.
Hinteregger, H. E., EUV flux variation during end of solar cycle 20 and beginning of cycle 21, observed from AE-C satellite, Geophys. Res. Lett., 4, 231, 1977.

Meriwether, J. W., Jr., D. G. Torr, and J. C. G. Walker, The $\mathrm{O}^{+}(2 \mathrm{P})$ emission at $7320 \AA$ in twilight, J. Geophys. Res., 83, 3311, 1978.

Noxon, J. F., and R. B. Norton, Changes in thermospheric composition inferred from twilight $\mathrm{O}^{+}\left({ }^{2} \mathrm{P}\right)$ emission, Planet. Space Sci., 27, 653, 1979.

McDade, I.C., W. E. Sharp, P. G. Richards, and D. G. Torr, On the inversion of $\mathrm{O}^{+}(2 \mathrm{D}-2 \mathrm{P}) 7320 \AA$ twilight airglow observations: A method for recovering both the ionization frequency and the thermospheric oxygen atom densities. J. Geophys. Res., 96, 259, 1991.
Press, W. H.,B. P. Flannery, S. A. Teukolsky, and W. T. Vetterling, Numerical Recipes: The Art of Scientific Computing, Cambridge University Press, New York, 1986.

Richards, P. G., and D. G. Torr, Ratios of photoelectron to EUV ionization rates for aeronomic studies, J. Geophys. Res., 93, 4060, 1988.

Rusch, D. W., D. G. Torr, P. B. Hays, M. R. Torr and A. O. Neir, Determination of the $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequency using satellite airglow and particle data and its implications on the EUV solar flux,Geophys. Res. Lett., 3, 537, 1976.

Rusch, D. W., D. G. Torr, and P. B. Hays, The O II (7319-7330 £) dayglow, J. Geophys. Res., 82, 719, 1977.

Russell, C. T., Geophysical coordinate transformations, Cosmic Electrodynamics 2, 184, 1971
Torr, D. G., The photochemistry of the upper atmosphere, in The Photochemistry of Atmospheres, edited by J. S. Levine, Academic Press, New York, pp. 165-278, 1985.

Torr, M. R., D. G. Torr, R. A. Ong, and H. E. Hinteregger, Ionization frequency for major thermospheric constituents as a function of solar cycle 21, Geophys. Res. Lett., 6,771, 1979.

Torr, M. R., D. G. Torr, P. G. Richards, and S. P. Yung, Mid and Low Latitude Model of Thermospheric Emissions: $1 . \mathrm{O}^{+}(2 \mathrm{P}) 7320 \AA$ and $\mathrm{N}_{2} 2 \mathrm{P} 3371 \AA$, J. Geophys. Res., In press 1990.
Yee, J. H. and V. J. Abreu, Exospheric temperatures deduced from 7320- to $7330-\AA$ $\left(\mathrm{O}^{+}(2 \mathrm{D})-\mathrm{O}^{+}(2 \mathrm{P})\right)$ twilight observations, J. Geophys. Res., 87, 913, 1982.


FIG. 1. (a) Sketch illustrating the idealized ground-based twilight observing geometry. The lines of sight corresponding to a number of ground-based observations made in the zenith and towards the setting sun are illustrated with the dashed and solid lines. The typical spatial distribution of the twilight $\mathrm{O}^{+(2 \mathrm{P})} 7320 \AA$ emission rates is shown with iso-emission contours which are drawn at logarithmic intervals. After McDade et al. (1991).


FIG. 1. (b) Same as Fig. 1a but illustrating how the twilight observations were made with the Visible Airglow Experiment on the AE-E satellite.


FIG. 2. The zenith and sunward twilight $7320 \AA$ column emission rates measured by the VAE channel 2 on the AE-E sunset pass of orbit 6855. The zenith emission rates (solid squares) were measured within $\pm 5$ degrees of the zenith; the sunward emission rates (open squares) were measured at elevation angles ranging from 35 to 45 degrees.


FIG. 3. The simulated ground-based twilight $\mathrm{O}^{+(2 \mathrm{P})} 7320 \AA$ observations discussed by McDade et al. (1991). The sunward column emission rates were calculated for an elevation of 20 degrees towards the azimuth of the setting sun.


FIG. 4. The channel 2 VAE $7320 \AA$ column emission rates measured within $\pm 5$ degrees of the zenith on AE-E orbit 6855 (data points) and the fit obtained using the parameters discussed in the text (solid line).


FIG. 5. The channel 2 VAE $7320 \AA$ column emission rates measured in the sunward direction at elevation angles between 35 and 45 degrees on orbit 6855 (data points) and the fit obtained using the parameters discussed in the text (solid line).


FIG. 6. The channel 2 VAE $7320 \AA$ column emission rates measured in the sunward direction at elevation angles between 75 and 85 degrees on orbit 6855 (data points) and the fit obtained using the parameters discussed in the text (solid line). N.B. these observations were not used to find the fitting parameters.


FIG. 7. The channel 2 VAE $7320 \AA$ column emission rates measured in the sunward direction at elevation angles between 65 and 75 degrees on orbit 6855 (data points) and the fit obtained using the parameters discussed in the text (solid line). N.B. these observations were not used to find the fitting parameters.


FIG. 8. The channel 2 VAE $7320 \AA$ column emission rates measured in the sunward direction at elevation angles between 55 and 65 degrees on orbit 6855 (data points) and the fit obtained using the parameters discussed in the text (solid line). N.B. these observations were not used to find the fitting parameters.


FIG. 9. The channel 2 VAE $7320 \AA$ column emission rates measured in the sunward direction at elevation angles between 45 and 55 degrees on orbit 6855 (data points) and the fit obtained using the parameters discussed in the text (solid line). N.B. these observations were not used to find the fitting parameters.


FIG. 10. The atomic oxygen density profile (solid line with error bars) reconstructed using the $\mathrm{H}_{\mathrm{O}}$ and $[\mathrm{O}]_{250}$ parameters obtained from the fit to the orbit 6855 twilight observations. The profile derived from the temperature and atomic oxygen densities measured by the NATE and OSS instruments is shown by the open square and dashed line. The atomic oxygen densities from the MSIS-86 model are represented by the dot-dashed curve.


FIG. 11. The channel 1 VAE $7320 \AA$ column emission rates measured within $\pm 5$ degrees of the zenith on AE-E orbit 7012 (data points). The emission rates were obtained by averaging the channel 1 counts over four 32 msec integration periods. The plotted data have been smoothed (in both the ordinate and abscissa) using a three point running average to illustrate the underlying trend.


FIG. 12. The orbit 7012 channel 1 VAE $7320 \AA$ column emission rates measured within $\pm 5$ degrees of the zenith and sunward in the elevation angle bands $35-45,45-55,55-$ $65,65-75$ and $75-85$ degrees. Each sequence of 50 points shows how the emission rates varied between the solar zenith angles 90 and 105 degrees. The plotted data have been smoothed as in Fig. 11 to illustrate the underlying trend. The solid line shows the unsmoothed fit obtained using the parameters discussed in the text. N.B. for the inversion described in the text raw unsmoothed data were used.


FIG. 13. The zenith and sunward twilight $7320 \AA$ column emission rates measured by the VAE channel 2 on the AE-E sunset pass of orbit 24564. The zenith emission rates (solid squares) were measured within $\pm 5$ degrees of the zenith; the sunward emission rates (open squares) were measured at elevation angles between 35 and 45 degrees.


FIG. 14. The channel 2 VAE $7320 \AA$ column emission rates measured within $\pm 5$ degrees of the zenith on AE-E orbit 24564 (data points) and the fit obtained using the parameters discussed in the text (solid line).


FIG. 15. The channel 2 VAE $7320 \AA$ column emission rates measured in the sunward direction at elevation angles between 35 and 45 degrees on orbit 24564 (data points) and the fit obtained using the parameters discussed in the text (solid line).


FIG. 16. The temperatures recovered in this work for AE-E orbits 6855, 7012 and 24564 (squares with error bars) compared with those deduced by Yee and Abreu (1982) from their analysis of zenith $7320 \AA$ twilight observations (solid triangles). The temperatures measured by the Neutral Atmosphere Temperature Experiment are shown by the open triangles highlighted with central dots on orbits 6855,7012 and 24564.


FIG. 17. The $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequencies recovered in this work for AE-E orbits 6855, 7012 and 24564 (squares with error bars) compared with the frequencies deduced by Abreu et al. (1980) from dayglow $7320 \AA$ observations (open triangles) and the frequencies calculated by Torr et al. (1979) from EUV flux measurements (solid triangles).


F10.7 FLUX

FIG. 18. The $\mathrm{O}^{+}(2 \mathrm{P})$ ionization frequencies recovered in this work for $\mathrm{AE}-\mathrm{E}$ orbits 6855,7012 and 24564 (squares with error bars) compared with the frequencies deduced by Abreu et al. (1980) from dayglow $7320 \AA$ observations (open triangles) and the frequencies calculated by Torr et al. (1979) from EUV flux measurements (solid triangles).

## Table Al

AE-E orbits with spinning satellite and VAE observations at $7320 \AA$


|  |  |  |  | 57008 | 59367 | 952.1 | 140.9 | 1098.9 | 2.9 | 7.88 | 13.56 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 75352 | 353 | 2 | N | 57008 | 18447 | 1042.1 | 141.2 | 1001.1 | 2.85 | 8.21 | 13.53 |
| 75354 | 372 | 1 | N | 16080 | 18447 | 1053.8 | 140.9 | 987.2 | 2.89 | 8.34 | 13.56 |
| 75354 | 378 | 2 | $\stackrel{N}{N}$ | 57712 | 5183 | 1023.8 | 140.8 | 1029.1 | 3.19 | 8.6 | 13.91 |
| 75356 | 395 | 1 | N | 2816 | ${ }_{1} 5183$ | 1022.6 | 141 | 1029.8 | 3.22 | 8.64 | 13.94 |
| 75356 | 397 | 2 | N | 16672 | 19039 | 1027.3 | 140.8 | 1023.3 | 3.26 | 8.78 | 13.98 |
| 75356 | 402 | 1 | N | 51304 | 53663 | 10272.9 | 140.8 | 1006.2 | 3.27 | 8.86 | 13.99 |
| 75356 | 406 | 1 | N | 78992 | 81359 | 1051.8 | 145.1 | 997 | 3.44 | 9.06 | 14.13 |
| 75358 | 420 | 1 | N | 3112 | 5479 10311 | 10661.9 | 145.1 | 981.6 | 3.42 | 9.1 | 14.11 |
| 75358 | 422 | 2 | N | 16944 | 19311 | 1067.1 | 145.1 | 932.9 | 3.4 | 9.21 | 14.02 |
| 75358 | 427 | 1 | N | 51512 | 53871 81495 | 1059 | 145 | 898.4 | 3.58 | 9.28 | 13.97 |
| 75358 | 431 | 1 | N | 79216 | 81495 | 1032.5 | 145 | 1009.7 | 3.91 | 9.48 | 14.55 |
| 75360 | 447 | 2 | N | 1708 | 9399 | 1019.6 | 144.9 | 1003 | 4.03 | 9.6 | 14.6 |
| 75360 | 452 | 1 | N | 51640 | 53999 81294 | 1019.6 | 144.7 | 641.8 | 5.76 | 9.64 | 13.4 |
| 75360 | 456 | 1 | N | 79656 | 81294 | 58723 | 145 | 1045.4 | 4.49 | 9.81 | 15.03 |
| 75362 | 470 | 1 | N | 3167 | 534 | 894.4 | 145.5 | 1136 | 5.23 | 10.17 | 15.78 |
| 75364 | 496 | 1 | N | 9735 | 102 | 386.5 | 145 | 1001.6 | 17.6 | 19 | 20.35 |
| 76022 | 798 | 1 | N | 84205 | 84836 | 386.5 | 144.4 | 985.8 | 10.01 | 15.09 | 20.66 |
| 76024 | 818 | 1 | N | 45492 | 47851 | 956 | 144.1 | 1150.6 | 12.48 | 17.02 | 22.8 |
| 76032 | 925 | 1 | N | 78820 | 81099 | 843.2 | 144.1 | 1054.2 | 12.7 | 17.78 | 23.29 |
| 76036 | 973 | 1 | N | 565 | 58955 | 869.2 | 144.4 | 1106.8 | 13.53 | 18.21 | 0.01 |
| 76038 | 1001 | 2 | N | 72171 | 74530 | 693.8 | 145.1 | 1195.6 | 14.29 | 18.53 | 0.83 |
| 76040 | 1026 | 2 | N | 67331 | 69698 | 893.8 | 140.8 | 974.5 | 13.93 | 18.85 | 0.5 |
| 76042 | 1048 | 2 | N | 41835 | 44202 | 875.5 | 141.1 | 1012.9 | 14.12 | 18.9 | 0.75 |
| 76042 | 1054 | 2 | N | 82043 | 84410 | 851.4 | 143.3 | 908.6 | 14.87 | 20.21 | 1.61 |
| 76048 | 1125 | 2 | N | 37843 | 40210 | 926.9 891.9 | 142.7 | 913.1 | 15.71 | 21.16 | 2.36 |
| 76052 | 1174 | 2 | N | 18387 | 20746 | 891.9 830.8 | 141.3 | 962.1 | 16.43 | 21.7 | 3.01 |
| 76054 | 1202 | 2 | N | 31466 | 33825 | 530.8 | 141.9 | 776.7 | 2.85 | 6.9 | 11.95 |
| 76093 | 1721 | 2 | N | 39296 | 41303 | 532.2 | 141 | 1871.3 | 10.66 | 14 | 18.04 |
| 76096 | 1764 | 2 | N | 50136 | 52135 | 543.1 | 139.6 | 723.4 | 4.15 | 8.17 | 13.46 |
| 76099 | 1801 | 2 | N | 25008 | 27015 | 519.9 | 139.4 | 731.6 | 4.4 | 8.33 | 13.68 |
| 76099 | 1810 | 2 | N | 82240 | 4239 | 619.9 | 137.1 | 530.5 | 5.66 | 10.77 | 14.83 |
| 76108 | 1931 | 2 | I | 69063 | 71062 | 667.7 | 137.9 | 927.7 | 8.63 | 11.39 | 17.63 |
| 76111 | 1967 | 2 | I | 36143 | 38150 | 317.7 | 137.9 | 438.3 | 6.83 | 12.07 | 15.86 |
| 76114 | 2008 | 2 | I | 325 | 345 | 5177 | 141.4 | 369.1 | 8.67 | 12.79 | 16.29 |
| 76117 | 2049 | 2 | I | 28815 | 30814 | 517.7 | 141.9 | 433.4 | 8.26 | 13.75 | 17.62 |
| 76120 | 2091 | 2 | I | 29974 | 31981 | 720.9 | 140.5 | 644.4 | 10.37 | 14.69 | 19.57 |
| 76123 | 2133 | 2 | I | 30958 | 32957 | 477.8 | 140.5 | 399.8 | 9.7 | 15.47 | 18.88 |
| 76126 | 2174 | 2 | N | 24230 | 26237 | 727.6 |  | 373.9 | 9.79 | 15.57 | 18.86 |
| 76126 | 2183 | 2 | N | 79582 | 81581 | 744.5 | 152.1 | 236.9 | 13.76 | 17.03 | 19.18 |
| 76132 | 2269 | 2 | I | 84238 | 85405 | 380.8 | 151.8 | 166.6 | 10.11 | 17.74 | 18.66 |
| 76135 | 2302 | 2 | I | 27158 | 29157 | 1146.5 | 151.8 |  |  |  |  |

* when an observing sequence spans more than one orbit only the first orbit number is listed

Table A1. continued


| 76135 | 2311 | 2 | I | 82309 | 84316 | 991.5 | 151 | 758.7 | 11.04 | 11.6 | 12.2 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 138 | 2344 | 2 | I | 26853 | 28852 | 654.2 | 151.7 | 450.6 | 13.28 | 18.65 | 22.57 |
| 76141 | 2387 | 1 | I | 32325 | 34324 | 395.7 | 151.5 | 725.4 | 15.9 | 19.49 | 0.89 |
| 76144 | 2426 | 1 | I | 12365 | 14364 | 553.2 | 152 | 543.5 | 15.61 | 20.17 | 0.62 |
| 76153 | 2557 | 1 | I | 39036 | 41035 | 458.3 | 156.6 | 639.5 | 18.57 | 22.77 | 3.84 3.92 |
| 76153 | 2563 | 2 | 1 | 75812 | 77811 | 460.3 | 156.9 | 636.9 | 18.67 | 22.89 23.5 | 0.49 |
| 76156 | 2594 | 1 | I | 6532 | 8235 | 923.1 | 155.9 | 155.7 | 16.16 | 23.67 | 23.75 |
| 76156 | 2600 | 2 | I | 43276 | 44979 | 1042.2 | 153.4 | 644.8 | 20.21 | 0.31 | 5.28 |
| 76159 | 2638 | 1 | I | 18500 | 20507 | 474.3 | 153.4 | 649.8 | 20.38 | 0.38 | 5.38 |
| 76159 | 2644 | 2 | I | 55372 | 57371 | 467.9 | 152.1 | 180.9 | 17.44 | 0.9 | 2.08 |
| 76162 | 2680 | 1 | I | 10716 | 12715 | 1107.1 | 151.9 | 152.3 | 16.74 | 1.06 | 1.2 |
| 76162 | 2689 | 2 | I | 65916 | 67923 | 1259.7 532.8 | 151.7 | 583.3 | 21.37 | 1.62 | 6.66 |
| 76165 | 2721 | 1 | I | 4684 | 6691 | 5338.4 | 151.6 | 387.7 | 20.95 | 2.62 | 6.25 |
| 76168 | 2768 | 1 | I | 27892 | 29899 | 738.4 672.4 | 151.6 | 369.7 | 21.39 | 2.75 | 6.23 |
| 76168 | 2774 | 2 | I | 64819 | 66714 | 672.4 | 151.6 | 570.3 | 22.81 | 3.52 | 8.06 |
| 76171 | 2809 | 1 | I | 21076 | 23075 | 667.1 | 151.4 | 457.3 | 22.16 | 3.63 | 7.48 |
| 76171 | 2815 | 2 | I | 57835 | 59842 | 667.1 | 153 | 653.7 | 1.12 | 5.01 | 10.22 |
| 76177 | 2896 | 1 | I | 44043 | 46050 | 472.7 484.3 | 153.3 | 629 | 1.16 | 5.09 | 10.21 |
| 76177 | 2902 | 2 | I | 80947 | 82946 | 484.3 | 153.1 | 453.8 | 0.74 | 5.7 | 9.83 |
| 76180 | 2937 | 1 | I | 36699 | 38698 | 678 | 155.2 | 444.6 | 0.73 | 5.82 | 9.9 |
| 76180 | 2943 | 2 | I | 73539 | 75546 | 7978 | 155.2 | 345.8 | 1.4 | 7.51 | 10.63 |
| 76186 | 3020 | 1 | I | 28450 | 30449 | 798.7 | 157 | 345.8 | 0.8 | 7.68 | 9.97 |
| 76186 | 3026 | 2 | I | 65178 | 67185 | 936.4 | 156.5 | 262.8 | 3.66 | 8.39 | 12.79 |
| 76189 | 3062 | 1 | I | 27898 | 29905 | 542.4 | 154.9 | 563.8 433.3 | 2.87 | 8.49 | 12.08 |
| 76189 | 3068 | 2 | I | 64570 | 66577 | 690.5 | 154.4 | 433.2 | 3.62 | 9.06 | 12.7 |
| 76192 | 3104 | 1 | I | 26818 | 28817 | 698.8 | 153.3 | 388 | 3.44 | 9.16 | 12.49 |
| 76192 | 3110 | 2 | I | 63658 | 65665 | 751.5 | 152.9 | 388.9 464 | 8.82 | 13.79 | 17.83 |
| 76210 | 3358 | 1 | I | 27049 | 29048 | 638.4 | 153.4 | 446.7 | 8.83 | 13.9 | 17.83 |
| 76210 | 3364 | 2 | I | 63825 | 65832 | 657.1 | 155.6 | 653.4 | 10.8 | 14.55 | 20.02 |
| 76213 | 3400 | 1 | I | 25625 | 27624 | 442.9 | 155.6 | 660 | 10.88 | 14.65 | 20.18 |
| 76213 | 3406 | 2 | 1 | 62377 | 64384 | 445.7 | 1569 | 195.1 | 10.33 | 15.44 | 16.95 |
| 76216 | 3442 | 1 | I | 23288 | 24711 | 632.4 | 156.9 | 461.7 | 10.47 | 15.62 | 19.76 |
| 76216 | 3448 | 2 | I | 60001 | 2007 | 625.1 | 157.2 | 758.8 | 13.04 | 16.45 | 22.19 |
| 76219 | 3485 | 1 | I | 27296 | 29295 | 343.7 | 145.7 | 795 | 13.42 | 16.61 | 22.5 |
| 76219 | 3491 | 2 | I | 63952 | 65959 | 316.7 | 145.7 | 467.4 | 12.39 | 17.34 | 21.31 |
| 76222 | 3529 | 1 | I | 35936 | 37895 | 557.7 527 | 143.5 143.2 | 467.4 524.2 | 12.73 | 17.44 | 21.8 |
| 76222 | 3535 | 2 | I | 72520 | 74519 | 527.7 | 141.3 | 544.6 | 14.35 | 17.99 | 23.38 |
| 76225 | 3571 | 1 | I | 26320 | 283 | 405.6 | 141.3 | 736.7 | 15 | 18.09 | 0.1 |
| 76225 | 3577 | 2 | I | 62872 | 64879 | 338.2 | 140.6 |  | 13.46 | 18.74 | 22.59 |
| 76228 | 3613 | 1 | I | 21464 | 23471 | 658.9 |  | 346.8 | 13.33 | 18.85 | 22.36 |
| 76228 | 3619 | 2 | I | 57720 | 59727 | 698.7 | 140.8 | 346.8 604.4 | 15.67 | 19.64 | 1.06 |
| 76231 | 3656 | 1 | I | 22632 | 24639 | 417.8 | 140.3 | 604.4 630.1 | 15.95 | 19.76 | 1.33 |
| 76231 | 3662 | 2 | I | 58888 | 60887 | 393.6 | 140.6 | 389.8 | 15.12 | 20.68 | 0.43 |
| 76234 | 3699 | 1 | I | 22280 | 24279 | 610.8 | 140.4 | 389.8 | 15.12 |  |  |

Table Al. continued

| Date | Orbit | Ch | Spin | Time (sec UT) | Altitude (km) |  |  | Local Solar Time (hr) |  |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| yyddd | $\#$ | $\#$ | type | on | off | on | midap off | on | mid |


| 76234 | 3705 | 2 | I | 58767 | 60390 | 378.9 | 140.6 | 400.8 | 16.87 | 20.77 | 0.62 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 76237 | 3739 | 1 | I | 3823 | 5830 | 372.8 | 139.7 | 627.8 | 17.66 | 21.5 | 2.78 |
| 76243 | 3832 | 1 | N | 41103 | 43102 | 268.3 | 141.4 | 707.6 | 20.5 | 23.13 | 5.67 |
| 76243 | 3838 | 2 | N | 76839 | 78846 | 245.1 | 142 | 750.1 | 20.87 | 23.32 | 6.09 |
| 76246 | 3875 | 1 | N | 38311 | 38622 | 244.3 | 142 | 375.1 | 2.79 | 3.5 | 4.19 |
| 76249 | 3926 | 1 | N | 79423 | 81430 | 386.4 | 142.7 | 528.4 | 21.12 | 1.36 | 6.41 |
| 76255 | 4006 | 1 | N | 26038 | 28045 | 442.2 | 137.8 | 438.1 | 22.38 | 2.91 | 88 |
| 76255 | 4012 | 2 | N | 61150 | 63157 | 488.2 | 137.3 | 390.7 | 22.12 | 3.02 | 7.21 |
| 76264 | 4138 | 1 | I | 17510 | 19517 | 433.4 | 135.3 | 363.4 | 0.68 | 5.83 | 10.05 |
| 76264 | 4144 | 2 | I | 52246 | 54253 | 389.9 | 134.7 | 394.5 | 1.19 | 5.97 | 10.47 |
| 76271 | 4248 | 1 | I | 42701 | 44700 | 447.4 | 136 | 273.7 | 2.61 | 8.01 | 11.69 |
| 76274 | 4290 | 1 | I | 27157 | 29164 | 393.1 | 142.3 | 297.4 | 3.95 | 9.02 | 13.24 |
| 76274 | 4299 | 2 | I | 77893 | 79900 | 428.3 | 141.9 | 262.8 | 3.69 | 9.25 | 2.97 |
| 76277 | 4338 | 1 | I | 32757 | 34508 | 414.6 | 141.7 | 192.3 | 4.42 | 10.3 | 12.65 |
| 76277 | 4347 | 2 | I | 83101 | 85108 | 481.4 | 140.6 | 197.1 | 3.76 | 10.55 | 12.99 |
| 76283 | 4433 | 1 | N | 39501 | 41508 | 349.8 | 139.8 | 51.6 | 7.02 | 12.31 | 16.09 |
| 76283 | 4439 | 2 | N | 72788 | 74723 | 351.2 | 139.2 | 228.7 | 7.1 | 12.43 | 81 |
| 76286 | 4478 | 1 | N | 29604 | 31603 | 275.7 | 137.9 | 281.2 | 8.9 | 13.21 | 18.12 |
| 76286 | 4486 | 2 | N | 73452 | 75451 | 342.2 | 137.5 | 219.7 | 7.9 | 13.45 | 17.1 |
| 76292 | 4573 | 1 | N | 32460 | 34467 | 233.2 | 142.8 | 266.7 | 11.34 | 15.72 | 20.45 |
| 76292 | 4579 | 2 | N | 65068 | 67075 | 250.1 | 141.8 | 246.9 | 10 | 15.83 | 20.16 |
| 76295 | 4625 | 1 | N | 55756 | 57763 | 273.4 | 148.4 | 207.7 | 11.29 | 16.7 | 20.27 |
| 76298 | 4675 | 1 | N | 67691 | 69690 | 246.1 | 161.5 | 229.1 | 13.16 | 17.83 | 22.35 |
| 76301 | 4721 | 1 | N | 57939 | 59090 | 168.3 | 161.4 | 211.3 | 17.71 | 19.17 | 23.16 |
| 76304 | 4773 | 1 | N | 78403 | 80402 | 258.7 | 169.8 | 204.1 | 14.62 | 20.63 | 23.82 |
| 76319 | 5008 | 1 | N | 37514 | 39521 | 241.8 | 184.1 | 190.3 | 18.65 | 1.84 | 3.73 |
| 76319 | 5014 | 2 | N | 69466 | 71473 | 246 | 184.1 | 187.2 | 18.3 | 1.95 | 3.31 |
| 76322 | 5057 | 1 | N | 41202 | 43201 | 243.3 | 238.8 | 247.5 | 23.18 | 2.32 | 8.25 |
| 76322 | 5063 | 2 | N | 73154 | 75161 | 247.2 | 239.1 | 244.2 | 22.11 | 2.54 | 7.1 |
| 76340 | 5340 | 1 | N | 6849 | 12152 | 250 | 247.2 | 250.2 | 0.38 | 9.02 | . 14 |
| 76346 | 5440 | 2 | N | 24041 | 29344 | 244.4 | 241.6 | 244.6 | 20.81 | 6.77 | 20.6 |
| 77003 | 5810 | 2 | N | 22542 | 27901 | 249.6 | 253.5 | 249.6 | 9.36 | 21.3 | 9.3 |
| 77007 | 5874 | 1 | N | 19054 | 19917 | 247.8 | 246.2 | 245.8 | 3.32 | 5.09 | 6.96 |
| 77007 | 5878 | 2 | N | 40502 | 41341 | 247.2 | 245.7 | 245.5 | 3.26 | 4.97 | 6.78 |
| 77007 | 5884 | 1 | N | 72630 | 73493 | 247.2 | 245.4 | 244.8 | 3 | 4.77 | 6.65 |
| 77009 | 5912 | 2 | N | 50094 | 55461 | 244.5 | 247.3 | 244.4 | 3.21 | 15.23 | 3.23 |
| 77016 | 6019 | 2 | N | 18533 | 19396 | 258.6 | 256.9 | 255.9 | 20.58 | 22.38 | 0.22 |
| 77016 | 6025 | 1 | N | 50614 | 51477 | 259.4 | 257.3 | 255.8 | 19.81 | 21.64 | 23.47 |
| 77016 | 6029 | 2 | N | 72006 | 72869 | 259.8 | 257.7 | 256.1 | 19.3 | 21.14 | 22.97 |
| 77034 | 6307 | 2 | N | 7516 | 8379 | 263.7 | 262.1 | 260.4 | 7.91 | 9.79 | 11.58 |
| 77034 | 6313 | 1 | N | 39636 | 40499 | 263.8 | 262.4 | 260.6 | 7.15 | 9.06 | 10.93 |
| 77034 | 6317 | 2 | N | 61044 | 61907 | 263.4 | 262.1 | 260.4 | 6.62 | 8.55 | 10.44 |
| 77036 | 6352 | 1 | N | 76996 | 82355 | 259 | 257.9 | 259.2 | 8.86 | 20.78 | 8.7 |
| 77040 | 6415 | 2 | N | 68508 | 72347 | 259.5 | 254 | 255.5 | 1.62 | 10.04 | 18.78 |

Table A1. continued
$\begin{array}{lcccc}\text { Date } & \text { Orbit } & \text { Ch } & \text { Spin } & \text { Time (sec UT) } \\ \text { yyddd } & \# & \# & \text { type } & \text { on }\end{array}$ Altitude (km) Local Solar Time (hr) yyddd \# \# type on off on mid/ap off on mid off

| 77043 | 6455 | 1 | N | 20716 | 21435 | 252 | 252.5 | 252.9 | 9.81 | 11.48 | 13.18 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 77043 | 6459 | 2 | N | 41916 | 42779 | 251.5 | 252.2 | 252.9 | 8.5 | 10.4 | 12.45 |
| 77047 | 6524 | 2 | N | 49619 | 54978 | 247.8 | 250.6 | 247.7 | 3.73 | 15.74 | 3.74 4.32 |
| 77052 | 6598 | 1 | N | 8483 | 9346 | 244.7 | 243.1 | 242.5 | 0.55 | 2.4 | 4.32 |
| 77052 | 6602 | 2 | N | 29907 | 30770 | 244 | 242.5 | 242.2 | 0.44 | 2.29 | 3.21 |
| 77052 | 6612 | 2 | N | 83467 | 84330 | 243.7 | 242.1 | 241.5 | 17.54 | 5.46 | 17.43 |
| 77063 | 6776 | 1 | I | 19554 | 24921 | 261.7 | 261.8 | 258.3 | 14.65 | 2.63 | 14.61 |
| 77068 | 6855 | 2 | I | 11962 | 17329 | 258.3 | 253.9 | 250.4 | 17.22 | 1.27 | 9.65 |
| 77077 | 7011 | 1 | N | 73225 | 76944 | 251.5 | 249.9 | 245.3 | 7.95 | 19.9 | 7.85 |
| 77081 | 7074 | 2 | N | 63705 | 69073 53376 | 245 | 249.6 | 240 | 2.52 | 14.49 | 2.51 |
| 77087 | 7168 | 1 | N | 48009 | 53376 | 240 | 241.6 258.8 | 261 | 14.3 | 22.24 | 5.95 |
| 77108 | 7297 | 1 | I | 2224 | 5767 | 258.8 | 263.2 | 264.9 | 19.3 | 7.21 | 19.17 |
| 77100 | 7372 | 2 | I | 21824 | 27183 | 264.8 | 258.4 | 258.7 | 14.3 | 2.29 | 14.26 |
| 77108 | 7497 | 1 | I | 2224 | 7591 | 258.8 | 273.3 | 277.4 | 19.16 | 7.03 | 18.97 |
| 77132 | 7895 | 1 | I | 60470 | 65837 | 277.4 | 273.3 | 270.2 | 20.62 | 8.51 | 20.39 |
| 77138 | 7980 | 2 | I | 90 | 6349 | 270.2 | 271 | 269.1 | 17.23 | 5.2 | 17.14 |
| 77142 | 8055 | 1 | I | 59013 | 64381 | 26 | 264.7 | 267.9 | 14.29 | 2.2 | 14.15 |
| 77146 | 8122 | 2 | I | 73821 | 79180 | . 9 | 264.7 | 261.5 | 9.71 | 21.62 | 9.59 |
| 77155 | 8257 | 1 | I | 22204 | 27571 | 261.5 | 261.8 | 252.9 | 0.66 | 12.61 | 0.59 |
| 77169 | 8492 | 2 | I | 74108 | 79475 | 252.8 | 282.1 | 281.8 | 20.34 | 8.2 | 20.12 |
| 77176 | 8600 | 1 | I | 50307 | 55674 | 281.6 |  | 279.4 | 17.95 | 5.82 | 17.74 |
| 77181 | 8672 | 2 | I | 7003 | 12370 | 279.3 | 280.9 | 279.4 | 14.28 | 1.85 | 13.45 |
| 77187 | 8770 | 1 | I | 17282 | 22498 | 278.3 | 276.4 |  | 9.67 | 17.58 | 1.33 |
| 77194 | 8882 | 2 | I | 16242 | 19793 | 275.8 | 274.6 | 276.9 | 9.67 | 9.38 | 21.34 |
| 77214 | 9211 | 1 | I | 59721 | 65088 | 266 | 263.2 | 266.8 |  | 20.91 | 23.45 |
| 77218 | 9279 | 2 | I | 79641 | 80768 | 64.6 | 263 | 263.2 | 11.55 | 23.2 | 11.1 |
| 77224 | 9366 | 1 | I | 28704 | 34071 | 288.6 | 282.7 | 288.7 284.9 | 11.35 | 22.91 | 10.83 |
| 77228 | 9436 | 2 | I | 61616 | 66975 | 284.8 | 2851 | 253.9 | 8.34 | 13.4 | 18.74 |
| 77275 | 10191 | 1 | N | 65941 | 68260 | 251.4 | 252.2 | 253.1 | 18.81 | 20.15 | 21.48 |
| 77275 | 10191 | 1 | N | 68269 | 68908 | 253.3 | 253.5 | 280.6 | 3.63 | 15.42 | 3.14 |
| 77279 | 10249 | 2 | N | 31181 | 36452 | 278.8 | 254.8 | 278.3 | 2.29 | 4.3 | 6.39 |
| 77285 | 10345 | 1 | N | 31332 | 32299 | 279.6 | 278.2 | 278.3 | 7.97 | 19.9 | 7.82 |
| 77314 | 10816 | 1 | N | 60643 | 66002 | 269.6 | 268.1 | 270.3 | 7.97 | 7.9 | 17.77 |
| 77318 | 10880 | 2 | N | 58186 | 62881 | 281.3 | 278.6 | 279.1 | 2.77 | 4.73 | 6.78 |
| 77324 | 10979 | 1 | N | 75834 | 76769 | 274.7 | 273.6 | 273.6 | 0.27 | 8.9 | 17.33 |
| 77328 | 11041 | 2 | N | 64298 | 68153 | 272.7 | 271.3 | 278.3 280.3 | 10.89 | 12.08 | 13.37 |
| 77350 | 11396 | 1 | N | 73536 | 74127 | 281 | 280.4 | 280.3 279.1 | 18.34 | 6.19 | 18.12 |
| 77354 | 11456 | 2 | N | 48472 | 53839 | 279 | 280.8 | 275.7 | 20.32 | 0.44 | 4.41 |
| 77360 | 11550 | 1 | N | 43752 | 45543 | 279.9 | 280.1 | 274.1 | 1.82 | 13.71 | 1.64 |
| 77364 | 11617 | 2 | N | 55663 | 61030 | 273.9 | 273.7 | 274.1 |  |  |  |
| 78005 | 11705 | 1 | N | 11190 | 13109 | 270.3 | 268 | 267.5 | 22.69 | 2.55 | 6.82 |
| 78054 | 12026 | 2 | N | 31466 | 33825 | 830.8 | 141.3 | 962.1 | 16.43 | 21.7 | 3.01 |
| 78028 | 12085 | 1 | N | 78389 | 83748 | 320.6 | 324.8 | 321 | 8.53 | 20.35 | 8.16 |
| 78032 | 12144 | 2 | N | 54260 | 56899 | 320.3 | 319.3 | 322.4 | 6.61 | 12.41 | 18.25 |

Table A1. continued


|  |  |  | I | 74599 | 78094 | 453.6 | 457.7 | 457.6 | 4.62 | 11.95 | 19.64 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 79096 | 18868 |  | I | 71815 | 75350 | 452.6 | 459.8 | 454.9 | 4.94 | 12.29 | 19.95 |
| 79102 | 18960 |  | I | 73247 | 76598 | 452.3 | 459.9 | 454.9 | 5.19 4.81 | ${ }_{9} 12.15$ | 19.29 14.85 |
| 79104 | 18992 | 1 | I | 74535 | 76886 | 452.1 | 457.8 | 458.7 | 4.81 | 13.64 13.05 | 20.86 |
| 79106 | 19023 | 1 | I | 75959 | 79702 | 452.6 | 458 | 456 | 5.21 | 9.85 | 14.51 |
| 79108 | 19054 | 1 | I | 77391 | 79542 | 453.8 | 458.4 | 454.6 | 4.04 | 11.86 | 19.38 |
| 79110 | 19085 | 1 | I | 78478 | 82109 | 454.7 | 454.7 | 454.7 | 4.06 | 12 | 19.69 |
| 79112 | 19115 | 1 | I | 74214 | 77837 | 454.7 455.8 | 453.4 | 454.4 | 4.04 | 11.92 | 19.73 |
| 79114 | 19146 | 1 | I | 75550 | 79173 | 455.8 | 452.2 | 454.1 | 4.09 | 11.81 | 19.71 |
| 79116 | 19177 | 1 | I | 76878 | 80493 | 452.2 | 449.5 | 457.1 | 3.66 | 11.62 | 19.2 |
| 79132 | 19424 | 1 | I | 13 | 79636 | 452.1 | 449.7 | 456.8 | 3.47 | 11.43 | 19.1 |
| 79134 | 19455 | 1 | I | 77261 | 80884 | 452.1 | 454.7 | 456.1 | 13.68 | 16.24 | 18.95 |
| 79136 | 19486 | 1 | I | 78501 | 82124 | 451.8 | 454.7 | 455 | 3.24 | 11.02 | 18.88 |
| 79138 | 19517 | 1 | I | 79741 | 83364 | 452.1 | 453 | 452 | 3.21 | 10.9 | 18.98 |
| 79142 | 19578 | 1 | I | 76620 | 7 | 451.4 | 455.2 | 449.5 | 3.48 | 11.31 | 19.33 |
| 79146 | 19640 | 1 | I | 79180 | 82923 | 451.4 | 455.5 | 449.2 | 3.66 | 11.59 | 19.49 |
| 79148 | 19670 | 1 | I | 74860 | 78603 | 451.2 | 455 | 449.1 | 3.85 | 11.93 | 19.75 |
| 79150 | 19701 | 1 | I | 76164 | 79907 | 451.4 | 455 | 449 | 4.07 | 11.99 | 19.56 |
| 79152 | 19731 | 1 | I | 71860 | 75483 | 452.3 | 453 | 449 | 4.13 | 12.08 | 19.81 |
| 79154 | 19762 | 1 | I | 73124 | 76755 | 453.3 | 451.9 | 448.8 | 4.25 | 12.09 | 19.87 |
| 79156 | 19793 | 1 | I | 74371 | 77994 | 454.3 | 451.1 | 448.5 | 4.24 | 12.05 | 19.94 |
| 79158 | 19824 | 1 | I | 75595 | 79218 | 454.9 | 450.6 | 448.4 | 4.35 | 11.97 | 19.93 |
| 79160 | 19855 | 1 | I | 76819 | 80442 | 454.9 | 450.2 | 448.6 | 4.35 | 11.88 | 19.84 |
| 79162 | 19886 | 1 | I | 78019 | 81642 | 454.5 | 450.2 | 449.1 | 4.29 | 11.85 | 19.75 |
| 79164 | 19917 | 1 | I | 79211 | 82834 | 453.4 | 448.8 | 455.2 | 3.25 | 11.22 | 18.83 |
| 79176 | 20103 | 1 | I | 80578 | 84201 | 447.1 | 448.8 | 453.7 | 3.05 | 9.44 | 15.37 |
| 79178 | 20133 | 1 | I | 76106 | 78977 | 4537 | 445.8 | 455.4 | 15.37 | 17.01 | 18.7 |
| 79178 | 20133 | 1 | I | 78986 | 79729 | 453.7 | 448.9 | 454.5 | 3.08 | 10.86 | 18.72 |
| 79181 | 20180 | 1 | I | 80673 | 84297 | 447.3 | 448.9 | 453.2 | 3.04 | 10.73 | 18.68 |
| 79184 | 20222 | 1 | I | 57154 | 60785 | 47.1 | 4515 | 451.1 | 3.23 | 10.8 | 18.72 |
| 79187 | 20267 | 1 | I | 50513 | 54136 | 447.1 | 451.5 |  | 3.67 | 11.24 | 18.95 |
| 79190 | 20318 | 1 | 1 | 77601 | 81224 | 447.3 | 451.9 | 449.5 | 3.67 | 11.86 | 19.42 |
| 79193 | 20364 | 1 | I | 76633 | 56 | 448.1 | 448.8 | 449.2 | 4.31 | 12.21 | 19.85 |
| 79196 | 20410 | 1 | I | 75633 | 9256 | 450.1 | 446.9 | 448.5 | 4.37 | 12.25 | 20.08 |
| 79199 | 20457 | 1 | I | 80168 | 83791 | 452.2 | 446.9 | 447.8 | 4.27 | 11.94 | 19.89 |
| 79202 | 20503 | 1 | I | 79024 | 82647 | 453.4 | 445.9 | 447.5 | 4.16 | 11.76 | 19.72 |
| 79205 | 20549 | 1 | I | 77872 | 1503 | 453.5 | 4468 | 450.8 | 3.53 | 11.49 | 19.2 |
| 79220 | 20781 | 1 | I | 77711 | 813 | 7.2 | 444.7 | 450 | 3.38 | 11.23 | 19.09 |
| 79223 | 20827 | 1 | I | 76511 | 80134 | 446.9 | 4498 | 447 | 3.17 | 11.57 | 20.27 |
| 79226 | 20871 | 1 | I | 64055 | 68038 | 446.3 | 449.8 | 445.4 | 3.8 | 11.41 | 19.15 |
| 79232 | 20966 | 1 | I | 78639 | 82261 | 445.3 | 452.3 | 445.4 | 4.03 | 11.76 | 19.34 |
| 79235 | 21012 | 1 | I | 77510 | 81133 | 445.4 | 4502 | 445.5 | 4.1 | 11.98 | 19.57 |
| 79238 | 21059 | 1 | I | 81950 | 85573 | 446.3 |  | 449.8 | 16.31 | 17.78 | 19.3 |
| 79253 | 21290 | 1 | N | 81405 | 85028 | 447.6 | 4478 | 451.5 | 12.2 | 15.79 | 19.27 |
| 79256 | 21335 | 1 |  | 74565 | 78188 | 443.7 |  | 45.5 | 3.91 | 11.78 | 19.34 |
| 79259 | 21381 | 1 | N | 78933 | 82555 | 443.5 | 442.4 |  |  |  |  |

## Appendix 2

The following is a listing of the Fortran program used to extract $7320 \AA$ emission rates and related orbital quantities from the AE-E Visible Airglow Experiment data files.


+ FORM = 'UNFORMATTED')

```
PRINT*, 'Enter NREADMAX ?'
```

    READ*, NREADMAX
    PRINT*, 'Do you want \(G \& Z\) cor \((1=Y, 0=N) ? '\)
    READ*, IGZC
    PRINT', 'Enter channel \# (1 or 2) '
    READ*, ICH
    IF (ICH.eq. 1) GOTO 11
    PRINT*,'Enter ch1 to ch2 zenith angle correction (if any) ?'
    READ*, zacor
    PRINT'*, 'MIN channel 1 zenith angle ?'
    READ*, ZMIN \(\quad\) 'MAX channel 1 zenith angle ?'
    READ*, ' ZMAX
    TYPE 15 (' Enter name of output file containing observations :')
    ACCEPT 10, FSTR
    OPEN (UNIT=20, NAME=FSTR, TYPE='NEW')
    TYPE 16 ("Enter name of output file containing the vectors :')
    16 FORMAT (' Enter na
ACCEPT 10, F2STR
OPEN (UNIT $=21, ~ N A M E=F 2 S T R, ~ T Y P E=' N E W ') ~$
C Assign initial values:
$\mathrm{NS}=0$
NERR=0
NSQ=0
NATN=0
NAVG=0
$S T O=0$.
ISVIF=1
ISKIP $=0$
$C$ Get start, stop times, and averaging parameter:
35 TYPE 40
40 FORMAT (' Enter start, stop times in seconds (default=0,86400):')
ACCEPT *, JSTART, JSTOP
ISTART = JSTART*1000
IF (JSTOP .EQ. 0) JSTOP $=86400$
ISTOP $=$ JSTOP*1000
TYPE 45
FORMAT (" Enter Filter Wheel Position (10=all):')
FORMAT (' Ent
$C$ Position Vae Data File to start time, obtain header information and
$C$ check mode

CALL SETUP(1)
CALL INITV (4, ISTART,IDATE,IORB, ID, ISPINF, INF, MODE1,ITON,
> ITOFF, IDMF, MODE2, ITON2, ITOFF2)
MODE $=$ MODE1
IF (IDMF.EQ.1.AND.ISTART.GE.ITON2) MODE=MODE2
NSAT $=1$
IF (ID . EQ. 'D') NSAT=2
IF (ID .EQ. 'E') NSAT=3

C Read the Vae Data File, bin error-free channel one data,
$C$ convert to Rayleighs, and store in arrays:
DO 290 NREAD $=1$, NREADMAX
CALL VAERD
IF (IE .EQ. 'E ') GOTO 260

IF (JFW .NE. 10 . AND. JFW .NE. IFW) GOTO 290
IF (OADATA (9).LT. ZMIN) GOTO 290
IF (OADATA (9).GT. ZMAX)GOTO 290
IF (IEND4.EQ.1.OR.ITIME.GT.ISTOP) STOP
IF (ICH.EQ.2)GOTO 241
IF (ISQ11.EQ.1.OR.ISQ12.EQ.1)GOTO 270
IF (ISQ11L.EQ.1.OR.ISQ12L.EQ.1) GOTO 270
IF (IATN1.EQ.1) GOTO 280
GOTO 242
CONTINUE
IF (ISQ2.EQ.1) GOTO 270
IF (IATN2.EQ.1)GOTO 280
CONTINUE
CALL RAYL
ISKIP=1
IF (ICE.NE.1) GOTO 290
CALL SUNVEC (IDATE, ITIME, SUN, GST)
sth=oadata (3)
sph=oadata (4)
if (oadata(4), 1t . 0.0) $\mathrm{sph}=360.0+$ oadata (4)
gstr=gst/57.29578
Tmat ( 1,1 ) $=\operatorname{Cos}$ (gstr)
Tmat (1, 2) $=-1.0 * \operatorname{SIN}$ (gstr)
Tmat ( 1,3 ) $=0.0$
Tmat ( 2,1 ) $=$ SIN (gstr)
Tmat ( 2,2 ) $=$ Cos (gstr)
Tmat $(2,3)=0.0$
Tmat $(3,1)=0.0$
Tmat $(3,2)=0.0$
Tmat $(3,3)=1.0$
CALL CART (sth, sph, sgeo)
CALL MAXMUL1(Tmat, sgeo,sgei)
sgei (1) $=$ sgei $(1)^{*}$ (oadata $\left.(1)+6370.0\right)$
$\operatorname{sgei}(2)=\operatorname{sgei}(2) *$ (oadata(1) +6370.0 )
sgei $(3)=$ sgei $(3) *$ (oadata $(1)+6370.0)$
IF (ICH.EQ.2) GOTO 245
$\operatorname{AV}=(\operatorname{Ray} 1(1)+\operatorname{Ray} 1(2)+\operatorname{Ray} 1(3)+\operatorname{Ray} 1(4)) / 4.0$
Bright =Av
Sig=SQRT (1.0* (ICH11+ICH12+ICH11L+ICH12L)) *STV(1)/4.0
za=ABS (OADATA (9))
GOTO 246
Bright=RAY2
Sig=SQRT (ABS (1.0*ICH2))*STV (2)
$\mathrm{za}=\mathrm{ABS}$ (OADATA (9) + zacor)
WRITE $(20,251)$ ITIME, Bright, Sig, OADATA (1), za, OADATA (6)
IF (ICH.EQ.1) WRITE $(21,250)$ ITIME, sgei (1), sgei (2), sgei (3).
$>\operatorname{sun}(1), \operatorname{sun}(2), \operatorname{sun}(3), \operatorname{V1L}(1), \operatorname{V1L}(2)$, V1L (3)
IF (ICH.EQ.2) WRITE $(21,250)$ ITIME, sgei (1), sgei (2), sgei (3),
$>\operatorname{sun}(1), \operatorname{sun}(2), \operatorname{sun}(3), \operatorname{v} 2 \mathrm{~L}(1), \mathrm{V} 2 \mathrm{~L}(2), \mathrm{V} 2 \mathrm{~L}(3)$
PRINT*, ITIME, Bright, Sig, OADATA (1), za, OADATA(6)
250 FORMAT (1X,I8,9E13.5)
251 FORMAT (1X, 18,' ',1E13.5,' ',1E13.5,' ',
> 1E13.5,' ',1E13.5,' ',1E13.5)
GOTO 290
NERR = NERR + 1
GOTO 290
$\mathrm{NSQ}=\mathrm{NSQ}+1$
GOTO 290
NATN $=$ NATN +1
GOTO 290
CONTINUE

## STOP

END
SUBROUTINE VAERD
$C$ This version reads VAX formatted VAE data.
C CALLING PROGRAM MUST PROVIDE COMMON /CVAERD/ AND PRESET IEND4 C TO ZERO. IEND4 IS RESET TO 1 UPON EOF ON VDF. IF CALLING ROUTINE C DOES NOT CHECK THIS IT MAY GO INTO AN INFINITE LOOP UPON EOF.

DIMENSION REC (295), IREC (295), $\operatorname{IBUFF}(295), \operatorname{BUFF}(295), \operatorname{OADATA}(22)$,
$\gg \quad$ OADATL (22), $, \mathrm{A} 1(3), \mathrm{A} 2(3), \mathrm{B} 2(3), \mathrm{E}(3), \mathrm{F}(3), \mathrm{P}(3), \mathrm{DF} 1(3), \mathrm{DF} 2(3)$,
$>\quad \mathrm{R}(3), \mathrm{V} 1(3), \mathrm{V} 2(3), \mathrm{V} 1 \mathrm{~L}(3), \mathrm{V} 2 \mathrm{~L}(3), \mathrm{RMOON}(3), \operatorname{SUN}(3)$
LOGICAL*1 SATID(4), ID, SATID1, SATID2, SD1, SD2, NI1, NI2
LOGICAL* 4 MODE1, MODE2
CHARACTER*40 INDNAME
CHARACTER*4 IE
INTEGER*4 SHFTR
INTEGER*2 LEN
EQUIVALENCE (REC, IREC), (IBUFF, BUFF), (SATID, IREC (3))
COMMON /CVAERD/ITIME,THET, ICH11,ICH12,ISQ11,ISQ12,ITIMEL, THETL, $>$ IFW,IE, ICH11L, ICH12L, ISQ11L, ISQ12L, IATN1, ICH2, ISQ2, IATN2, $>$ TBAF1,TBAF2,TBAF11,TBAF22,TAEL,TBEL,TPMT1,TPMT2,TFW, IEND4, $>$ IX,ISVIF,OADATA,OADATL

COMMON /CVEC/ V1,V2,V1L,V2L,R,sun,gst
DATA P/0.,-.398,.917/, LORBIT/0/
DATA SAVE1/2.0/, PI/3.14159/, FAC/57.29578/
DATA ISW,IIREAD/2*0/, INUNIT/4/, IANGSW/0/, PI2/6.28318/
C READ HEADER AND FIRST RECORD UNLESS FILE IS PRE-POSITIONED USING
C ENTRY INITV
IF ( ISW .NE. 0) GO TO 100
$\operatorname{READ}(\operatorname{INUNIT}, E N D=250, E R R=250) \quad$ ( $\operatorname{IREC}(I), I=1,12$ )
IDATE $=$ IREC (1)
IORBIT = IREC (2)
ISF=1
READ (INUNIT , END $=250$, $\mathrm{ERR}=250$ ) (IBUFF ( $K$ ), $K=1,295$ )
ISW=1
100 IIREAD=IIREAD +1
IF (IIREAD.NE.1) GO TO 400
DO $120 \mathrm{~K}=1,295$
$120 \operatorname{IREC}(\mathrm{~K})=\operatorname{IBUFF}(\mathrm{K})$
$\operatorname{IF}(\operatorname{REC}(275) . G T \cdot 23.0$. AND. $\operatorname{BUFF}(275)$. LT . 1.0) $\operatorname{REC}(275)=\operatorname{REC}(275)-24.0$ $\operatorname{IF}(\operatorname{REC}(277) . G T .179 .0 . \operatorname{AND} \cdot \operatorname{BUFF}(277) . \operatorname{LT} \cdot-179.0) \operatorname{REC}(277)=\operatorname{REC}(277)-$ $>360.0$
IF (REC (277) .LT. -
$>179.0$. AND. $\operatorname{BUFF}(277) . \mathrm{GT} .179 .0) \operatorname{REC}(277)=\operatorname{REC}(277)+360.0$
$\operatorname{IF}(\operatorname{REC}(281) . \operatorname{GT} .179 .0 . \operatorname{AND} \cdot \operatorname{BUFF}(281) . \operatorname{LT} .-179.0) \operatorname{REC}(281)=\operatorname{REC}(281)-$ $>230.0$

READ (INUNIT, END $=250, \mathrm{ERR}=250$ ) (IBUFF (K), $\mathrm{K}=1,295$ )

C IF ORBIT IS SPINNING, FIND OA VECTORS NEEDED FOR INTERPOLATIONS
IAFLAG=1
IF (ISF.NE.1.OR.ISVIF.EQ.0) GOTO 260
DO $190 \mathrm{~K}=284,295$
190 IF (REC (K).LT. 1000 . OR. REC (K) . GT . 1000) IAFLAG $=0$
IF (IAFLAG.EQ.0) GOTO 260
IF (LORBIT.EQ.IORBIT)GOTO 200
LORBIT $=$ IORBIT
CALL SUNVEC (IDATE,ITIME,SUN,gst)
CALL CROSS (P,SUN,E)
200 CALL CART (REC (287), $\operatorname{REC}(286), \operatorname{A1})$
CALL CART (REC (293), $\operatorname{REC}(292), \operatorname{A2})$
CALL CART (BUFF (287), BUFF (286), B1)
CALL CART (BUFF (293), BUFF (292), B2)
DO $210 \mathrm{NN}=1,3$
$\mathrm{DF} 1(\mathrm{NN})=\mathrm{A} 1(\mathrm{NN})-\mathrm{B} 1(\mathrm{NN})$
$210 \mathrm{DF} 2(\mathrm{NN})=\mathrm{A} 2(\mathrm{NN})-\mathrm{B} 2(\mathrm{NN})$
CALL CROSS (DF2, DF1,R)
CALL DETER (A1, B1, R, DET1)
CALL DETER (A2, B2, R, DET2)
GAMMA $=$ PI 2-ANGLE (A1, B1)
CMCH1 $=\operatorname{COS}(\operatorname{REC}(285) / \mathrm{FAC})$
$\mathrm{CMCH} 2=\operatorname{COS}(\operatorname{REC}(291) / \mathrm{FAC})$
CALL LUNVEC (A1, A2, CMCH1, CMCH2, RMOON)
GO TO 260
C IF EOF, SET FLAG AND RETURN
250 IEND4 $=1$
GO TO 4
C UNPACK THE DATA RECORD
260 IF ( $\mathrm{REC}(15)$.LE. 1.0 ) GO TO 280
IF (SAVE1.NE. 2) GO TO 270
$\operatorname{REC}(16)=1.570681$
$\operatorname{REC}(17)=0.0$
$\operatorname{REC}(15)=1.0$
GO TO 290
$270 \quad \operatorname{REC}(15)=$ SAVE 1
GO TO 300
$280 \quad$ SAVE1 $=\operatorname{REC}(15)$
$290 \mathrm{DELTH}=.125$ * $\operatorname{REC}(17)$
THET1 $=\operatorname{REC}(16)-$ DELTH
THET2 $=\operatorname{REC}(16)+.0626 * \operatorname{REC}(17)-$ DELTH
300 ITIME $=\operatorname{IREC}(1)-125$
ITIMEL $=\operatorname{IREC}(1)-63$
TBAF1 $=\operatorname{REC}(3)$
TBAF2 $=$ REC (4)
TBAF11 = REC(5)
TBAF22 $=$ REC (6)
TAEL $=\operatorname{REC}(7)$
TBEL $=$ REC (8)
TPMT1 $=\operatorname{REC}(9)$
TPMT2 $=$ REC (10)
TFW $=\operatorname{REC}(11)$
$\mathrm{C} J J=$ INDEX $T O$ STATUS BITS

```
400 JJ = IIREAD * 4 + 14
    ITIME = ITIME + 125
    ITIMEL = ITIMEL + 125
    IF ( IREC(JJ+1) .NE. -1 ) GO TO 410
```

```
        ICH11 = -1
        ICH12 = -1
        GO TO 420
410 ICH11 = JIBITS (IREC (JJ +1),16,16)
        ICH12 = JIBITS (IREC (JJ +1),0,16)
        IF ( IREC (JJ+2) .NE. -1) GO TO 430
        ICH11 = -1
        ICH12 = -1
        GO TO 440
430 ICH11L = JIBITS (IREC (JJ+2),16,16)
    ICH12L = JIBITS (IREC (JJ +2),0,16)
    IF(IIREAD.NE.1) ICH2 = IREC(JJ+3)
    ITEMP = IREC(JJ)
C ZENITH ANGLE CALCULATIONS (OLD METHOD)
    THET = 0.0
    THETL = 0.0
    IF(THET1.GT.PI2 .OR. THET2.GT.PI2 .OR. REC(15).GT.1.0)GO TO 450
    IF(IANGSW .EQ. 1) GO TO 450
THET1 = THET1 + DELTH
THET = ACOS (COS (THET1) * REC(15)) * 57.29583
IF (THET1 .LT. 0.0) THET = 360.0 - THET
IF(THET1 .GE. 6.28318) THET1 = THET1 - 6.28318
IF (THET1 .GT.3.14159) THET = 360.0 - THET
THET2 = THET2 + DELTH
THETL = ACOS (COS(THET2) * REC(15)) * 57.29583
IF (THET2 .LT. 0.0) THETL = 360.0 - THETL
IF(THET2 .GE. 6.28318) THET2 = THET2 - 6.28318
IF (THET2 .GT. 3.14159) THETL = 360.0 - THETL
450 CONTINUE
C UNPACK STATUS WORD
```

```
IWORD=JIBITS (ITEMP, 31,1)
```

IWORD=JIBITS (ITEMP, 31,1)
IF ( IWORD.EQ. 0) GO TO 500
IF ( IWORD.EQ. 0) GO TO 500
IE='E '
IE='E '
GO TO 501
GO TO 501
500 IE='
501 ISQ2=JIBITS (ITEMP,24,1)
501 ISQ2=JIBITS (ITEMP,24,1)
ISQ11=JIBITS(ITEMP,19,1)
ISQ11=JIBITS(ITEMP,19,1)
ISQ12=JIBITS (ITEMP, 18,1)
ISQ12=JIBITS (ITEMP, 18,1)
ISQ11L=JIBITS (ITEMP,17,1)
ISQ11L=JIBITS (ITEMP,17,1)
ISQ12L=JIBITS (ITEMP,16,1)
ISQ12L=JIBITS (ITEMP,16,1)
IATN1=JIBITS (ITEMP,9,1)
IATN1=JIBITS (ITEMP,9,1)
IATN2 = JIBITS (ITEMP,8,1)
IATN2 = JIBITS (ITEMP,8,1)
IFW=JIBITS (ITEMP,0,3)
IFW=JIBITS (ITEMP,0,3)
C PERFORM OA INTERPOLATIONS

```
```

RAT=(IIREAD-1)/64.

```
RAT=(IIREAD-1)/64.
RATL=(IIREAD-.5)/64.
RATL=(IIREAD-.5)/64.
DO 700 N=1,8
DO 700 N=1,8
OADATA (N ) =REC (273+N})+(\operatorname{BUFF}(273+N)-\operatorname{REC}(273+N))*RAT
OADATA (N ) =REC (273+N})+(\operatorname{BUFF}(273+N)-\operatorname{REC}(273+N))*RAT
700
OADATL}(N)=REC (273+N)+(\operatorname{BUFF}(273+N)-REC (273+N))*RAT
OADATL}(N)=REC (273+N)+(\operatorname{BUFF}(273+N)-REC (273+N))*RAT
IF (OADATA (2).LT . 0.0) OADATA (2) =OADATA (2) +24.0
IF (OADATA (2).LT . 0.0) OADATA (2) =OADATA (2) +24.0
IF (OADATA (2).EQ.0.0)OADATA (2) =24.0
IF (OADATA (2).EQ.0.0)OADATA (2) =24.0
IF (OADATL (2).LT . 0.0) OADATL (2) =OADATL (2) +24.0
IF (OADATL (2).LT . 0.0) OADATL (2) =OADATL (2) +24.0
IF (OADATA (8).LT . -180.0)OADATA (8) =OADATA (8) +360.0
IF (OADATA (8).LT . -180.0)OADATA (8) =OADATA (8) +360.0
IF (OADATL (8).LT . -180.0)OADATL (8)=OADATL (8) +360.0
IF (OADATL (8).LT . -180.0)OADATL (8)=OADATL (8) +360.0
IF(ISF.EQ.1)GOTO 720
IF(ISF.EQ.1)GOTO 720
DO 710 N=9,22
```

DO 710 N=9,22

```
```

$\operatorname{OADATA}(N)=\operatorname{REC}(273+N)+(\operatorname{BUFF}(273+N)-\operatorname{REC}(273+N)) * \operatorname{RAT}$
$\operatorname{OADATL}(\mathrm{N})=\operatorname{REC}(273+\mathrm{N})+(\operatorname{BUFF}(273+\mathrm{N})-\operatorname{REC}(273+\mathrm{N})) * \operatorname{RATL}$
GOTO 800
$\operatorname{OADATA}(9)=\operatorname{REC}(282)+\operatorname{RAT} * 8 . * \operatorname{REC}(283)$
IF (OADATA (9) .GE . 180.) OADATA (9) =OADATA (9) -360.
IF (OADATA (9). LE -180.$)$ OADATA $(9)=O \operatorname{ADATA}(9)+360$.
OADATA (10) = REC (283)
OADATL (9) $=\operatorname{REC}(282)+\operatorname{RATL} * 8 . * \operatorname{REC}(283)$
IF (OADATL (9).GE . 180.) OADATL (9) =OADATL (9) -360.
IF (OADATL (9).LE . $\mathbf{- 1 8 0}$.) OADATL $(9)=\operatorname{OADATL}(9)+360$.
OADATL (10) = REC (283)
IF (ISVIF.EQ.0)GOTO 800
IF (IAFLAG.EQ.0) GOTO 750
DELTA=RAT*GAMMA
$\mathrm{CD}=\mathrm{COS}$ (DELTA)
$C E=C O S$ (GAMMA-DELTA)
CALL SOLVE (A1, B1, R, CD, CE, 0, DET1, V1)
CALL SPHERE (V1, OADATA (14), OADATA (13))
OADATA (11) =ANGLE (V1, SUN) *FAC
OADATA (12) =ANGLE (V1, RMOON) *FAC
CALL MULT (SUN, E, P,V1,F)
CALL SPHERE (F, OADATA (15) , OADATA (16))
CALL SOLVE (A2, B2, R, CD, CE, 0, DET2,V2)
CALL SPHERE (V2, OADATA (20), OADATA (19))
OADATA (17) =ANGLE (V2, SUN) *FAC
OADATA (18) =ANGLE (V2, RMOON) *FAC
CALL MULT (SUN, E, P, V2,F)
CALL SPHERE (F,OADATA (21), OADATA(22))
DELTAL=RATL*GAMMA
$\mathrm{CDL}=\mathrm{COS}$ (DELTAL)
CEL $=\operatorname{COS}$ (GAMMA-DELTAL)
CALL SOLVE (A1, B1, R, CDL, CEL, 0, DET1,V1L)
CALL SPHERE (V1L, OADATL (14), OADATL (13))
OADATL (11) =ANGLE (V1L, SUN) *FAC
OADATL (12) =ANGLE (V1L, RMOON) *FAC
CALL MULT (SUN, E, P, V1L, F)
CALL SPHERE (F, OADATL (15), OADATL (16))
CALL SOLVE (A2, B2, R, CDL, CEL, 0, DET2, V2L)
CALL SPHERE(V2L, OADATL (20), OADATL (19))
OADATL (17) =ANGLE (V2L, SUN) *FAC
OADATL (18) =ANGLE (V2L, RMOON) *FAC
CALL MULT (SUN, E, P,V2L,F)
CALL SPHERE (F, OADATL (21), OADATL (22))
GOTO 800
750 DO $755 \mathrm{~N}=11,22$
OADATA $(\mathrm{N})=-99999$.
$\operatorname{OADATL}(\mathrm{N})=-99999$.
C APPLY DEAD TIME CORRECTION

```
```

800 IF(ICH11 .LT. 0) GO TO 810

```
800 IF(ICH11 .LT. 0) GO TO 810
    ICH11=ICH11/(1-ICH11/1.11E5)
    ICH11=ICH11/(1-ICH11/1.11E5)
    ICH12=ICH12/(1-ICH12/1.11E5)
    ICH12=ICH12/(1-ICH12/1.11E5)
810 IF(ICH11L .LT. 0) GO TO 820
810 IF(ICH11L .LT. 0) GO TO 820
    ICH11L=ICH11L/(1-ICH11L/1.11E5)
    ICH11L=ICH11L/(1-ICH11L/1.11E5)
    ICH12L=ICH12L/(1-ICH12L/1.11E5)
    ICH12L=ICH12L/(1-ICH12L/1.11E5)
820 IF(ICH2 .LT. 0) GO TO 830
820 IF(ICH2 .LT. 0) GO TO 830
    ICH2=ICH2 / (1-ICH2/4.44E5)
    ICH2=ICH2 / (1-ICH2/4.44E5)
830 IF (IIREAD.EQ.64) IIREAD=0
830 IF (IIREAD.EQ.64) IIREAD=0
    IX=IIREAD
    IX=IIREAD
    4 RETURN
```

    4 RETURN
    ```
```

    ENTRY INITV(IUNTP,JTIMEP,IDDATE,IORB,ID,ISPINF,INF,
    >MODE1,ITON, ITOFF,IDMF,MODE2,ITON2,ITOFF2)
    INUNIT= IUNTP
    IIREAD = 0
    ISW = 0
    SAVE1 = 2.0
    JTIME = JTIMEP - 8000
    READ (INUNIT, END=4,ERR=4) (IREC (I), I=1,12)
    ID = SATID(1)
    IDDATE = IREC(1)
    IDATE=IDDATE
    IORB = IREC (2)
    IORBIT=IORB
    IF(ID.EQ.'C'.OR.ID.EQ.'C') INDNAME=
    +'SPRLC$DISK1:[VAECOMMON. IND]VDFC.DAT'
    IF(ID.EQ.'D'.OR.ID.EQ.'d') INDNAME=
    +'SPRLCSDISK1:[VAECOMMON. IND]VDFD.DAT'
    IF(ID.EQ.'E'.OR.ID.EQ.'e') INDNAME=
    +'SPRLC$DISK1:[VAECOMMON.IND]VDFE.DAT'
    OPEN(UNIT=9,NAME=INDNAME,STATUS='OLD',READONLY, ERR=179)
    GOTO 900
    179 TYPE *, 'ERROR OPEN'
READ (9,910,END=940) SATID1,IORB1,MODE1,SD1,NI1, IDATE1,
+ITON1, ITOFF1,MT1
FORMAT (A1, 1X, I5,1X, A4,1X,2A1, 1X, 3(I5,1X),I3)
IF (IORB1.NE.IORB) GOTO 900
IORB2=0
READ (9,910, END=920) SATID2,IORB2 ,MODE2,SD2,NI2,IDATE2,
+ITON2,ITOFF2,MT2
ISF=0
IF (SD1.EQ.'S') ISF=1
ISPINF=ISF
INF=0
IF (NII.EQ.'N') INF=1
ITON=ITON1
ITOFF=ITOFF1
IDMF=0
IF(IORB1.EQ.IORB2) THEN
IDMF=1
ELSE
MODE2=
ITON2=0
ITOFF2=0
ENDIF
CLOSE (UNIT=9)
GOTO 950
TYPE *,'ORBIT NOT FOUND IN INDEX'
CLOSE(UNIT=9)
STOP
CONT INUE
READ(INUNIT, END=4,ERR=4) (IBUFF (K), K=1,295)
ISW= ISW+1
IF(IBUFF(1) .LT. JTIME) GO TO 950
RETURN
ENTRY SETUP(IANGZ)
IANGSW= IANGZ
GO TO 4

```

END
FUNCTION ANGLE(V1,V2)
DIMENSION V1(1), V2 (1)
ANGLE \(=A \operatorname{COS}((\mathrm{~V} 1(1) * \mathrm{~V} 2(1)+\mathrm{V} 1(2) * \mathrm{~V} 2(2)+\mathrm{V} 1(3) * \mathrm{~V} 2(3)) /\)
\(>\operatorname{SQRT}((\mathrm{V} 1(1) * * 2+\mathrm{V} 1(2) * * 2+\mathrm{V} 1(3) * * 2) *(\mathrm{~V} 2(1) * * 2+\mathrm{V} 2(2) * * 2+\mathrm{V} 2(3) * * 2)))\)
RETURN
END
SUBROUTINE CROSS (A, B,C)
DIMENSION A(1), B(1), C(1)
\(\mathrm{C}(1)=\mathrm{A}(2) * \mathrm{~B}(3)-\mathrm{A}(3) * \mathrm{~B}(2)\)
\(\mathrm{C}(2)=\mathrm{A}(3) * \mathrm{~B}(1)-\mathrm{A}(1) * \mathrm{~B}(3)\)
\(\mathrm{C}(3)=\mathrm{A}(1) * \mathrm{~B}(2)-\mathrm{A}(2) * \mathrm{~B}(1)\)
\(\mathrm{CL}=\mathrm{SQRT}(\mathrm{C}(1) * * 2+\mathrm{C}(2) * * 2+\mathrm{C}(3) * * 2)\)
\(C(1)=C(1) / C L\)
\(C(2)=C(2) / C L\)
\(C(3)=C(3) / C L\)
RETURN
END
SUBROUTINE MULT (SUN, E, P, V, F)
DIMENSION SUN(1), E(1), P(1), V(1),F(1)
\(F(1)=\operatorname{SUN}(1) * V(1)+\operatorname{SUN}(2) \star V(2)+\operatorname{SUN}(3) * V(3)\)
\(F(2)=E(1) * V(1)+E(2) * V(2)+E(3) * V(3)\)
\(\mathrm{F}(3)=\mathrm{P}(1) * \mathrm{~V}(1)+\mathrm{P}(2) * \mathrm{~V}(2)+\mathrm{P}(3) * \mathrm{~V}(3)\)
RETURN
END
SUBROUTINE CART (THETA, PHI, W)
DIMENSION W(1)
T=THETA*. 01745329
\(\mathrm{P}=\mathrm{PHI}\) *. 01745329
\(\mathrm{W}(1)=\cos (\mathrm{T}) * \cos (\mathrm{P})\)
\(W(2)=\operatorname{COS}(T) * \operatorname{SIN}(P)\)
\(W(3)=S I N(T)\)
RETURN
END
SUBROUTINE SPHERE (V,TH, PH)
DIMENSION V(1)
\(\mathrm{AV}=\mathrm{SQRT}(\mathrm{V}(1) * * 2+\mathrm{V}(2) * * 2+\mathrm{V}(3) * * 2)\)
\(V(1)=V(1) / A V\)
\(V(2)=V(2) / A V\)
\(V(3)=V(3) / A V\)
\(\mathrm{PH}=\mathrm{ATAN} 2(\mathrm{~V}(2), \mathrm{V}(1)) * 57.29578\)
\(\mathrm{TH}=\mathrm{ASIN}(\mathrm{V}(3)) * 57.29578\)
RETURN
END
SUBROUTINE SOLVE ( \(X, Y, Z, C A X, C A Y, C A Z, D E T, V)\)
DIMENSION \(\mathrm{X}(1), \mathrm{Y}(1), \mathrm{Z}(1), \mathrm{V}(1)\)
IF (DET.EQ.0.0) GOTO 100
\(V(1)=(C A X *(Y(2) * Z(3)-Y(3) * Z(2))+X(2) *(Y(3) * C A Z-C A Y * Z(3))\)
\(>+X(3) *(C A Y * Z(2)-Y(2) * C A Z)) / D E T\)
\(V(2)=(X(1) *(C A Y * Z(3)-Y(3) * C A Z)+C A X *(Y(3) * Z(1)-Y(1) * Z(3))\)
\(>+X(3) *(Y(1) * C A Z-C A Y * Z(1))) / D E T\)
\(V(3)=(X(1) *(Y(2) * C A Z-C A Y * Z(2))+X(2) *(C A Y * Z(1)-Y(1) * C A Z)\)
\(>+\operatorname{CAX} *(Y(1) * Z(2)-Y(2) * Z(1))) / D E T\)
RETURN
\(100 \mathrm{~V}(1)=0\).
\(V(2)=0\).
\(V(3)=0\).
RETURN
```

END
SUBROUTINE DETER(X,Y,Z,DET)
DIMENSION X(1),Y(1),Z(1)
DET= X(1)* (Y(2)*Z(3)-Y(3)*Z(2)) + X(2)* (Y(3)*Z(1)-Y(1)*Z(3))
> + X(3)*(Y(1)*Z(2)-Y(2)*Z(1))
RETURN
END
SUBROUTINE LUNVEC(A1,A2,G,H,RMOON)
DIMENSION A1 (1),A2 (1),RMOON(1)
A=A1 (1)
B=A1 (2)
C=A1 (3)
D=A2(1)
E=A2 (2)
F=A2 (3)
S = (E*C-B*F)**2 + (A*F-D*C)**2 + (E*A-D*B)**2
Q = (A*H-D*G)* (F*A-D*C) + (E*G-B*H)* (E*C-B*F)
T = (E*G-B*H)**2 + (A*H-D*G)**2 - (E*A-D*B)**2
R = (2*Q)**2 - 4* S*T
IF(R.LT.0.)GOTO 100
RMOON(3) = (2*Q+SQRT(R)) / (2*S)
V = H-F*RMOON(3)
W = G-C*RMOON(3)
RMOON (1) = (W*E-V*B) / (E*A -D*B)
RMOON(2) = (V*A-D*W) / (E*A-D*B)
RETURN
100 RMOON (1) =0.
RMOON (2) =0.
RMOON (3) =0.
RETURN
END
SUBROUTINE SUNVEC(IDATE,ITIME,SUN,gst)
DIMENSION SUN(1)
DATA RAD/57.29578/
REAL*8 DJ,FDAY
FDAY=FLOAT (ITIME)/86400000.
IYR=IDATE/1000
IDAY=IDATE-IYR*1000
DJ=365*IYR+(IYR-1)/4+IDAY+FDAY-0.5D0
T=DJ/36525.
VL=DMOD (279.696678+.9856473354*DJ, 360.DO)
gst =dmod (279.690983+.9856473354*DJ +360.*FDAY+180.,360.D0)
G=DMOD (358.475845+.985600267*DJ,360.DO) /RAD
SLONG=VL+(1.91946-.004789*T)*SIN(G)+.020094*SIN (2 . *G)
OBLIQ= (23.45229-0.0130125*T)/RAD
SLP = (SLONG-.005686)/RAD
SIND=SIN(OBLIQ) *SIN(SLP)
COSD=SQRT(1.-SIND**2)
SDEC=ATAN (SIND/COSD)
SRASN=3.14159-ATAN2 (1/TAN (OBLIQ)*SIND/COSD, -COS (SLP) /COSD)
SUN (1)=COS (SRASN) *COS (SDEC)
SUN (2) =SIN (SRASN) * COS (SDEC)
SUN (3)=SIN (SDEC)
RETURN
END
SUBROUTINE RAYL
C SUBROUTINE RAYL CONVERTS RAW VAE COUNTS TO RAYLEIGHS, AND STORES C THE VALUES IN RAY1 (1:4) AND RAY2. THE CHANNEL NUMBER REQUESTED

```

C MUST BE IN ICH AND THE SATELLITE NUMBER ( \(\mathrm{C}=1, \mathrm{D}=2, \mathrm{E}=3\) ) MUST BE
C IN NSAT. FIRST, DARK COUNT IS ESTIMATED FROM THE
C PHOTOMULTIPLIER TUBE TEMPERATURE, THEN THE SENSITIVITY IS
C INTERPOLATED FOR THE SPECIFIED FILTER WHEEL TEMPERATURE, FILTER
C WHEEL POSITION, CHANNEL, AND SATELLITE.
C SUBROUTINE ALSO SUBTRACTS OUT THE GALACTIC AND ZODIACAL BACKGROUND C IF IGZC=1.
C CALLING PROGRAM MUST SUPPLY /CVAERD/ AND /CRAYL/ IN COMMON
C Modification \(6 / 15 / 88\) to work with galactic and zodiacal background \(C\) subtraction on SPRLC.
c * errors detected in VAEREAD on 1/7/91 fixed here * C Calling program must initialize ISKIP to 0 and then set it to 1 \(C\) immediately after RAYL is called for the first time.
```

INTEGER*2 LEN
INTEGER*4 RA (4),DEC (4), ELAT (4),ELON (4)
CHARACTER*4 IE
DIMENSION GAL (5, 120,60), ZOD (5,60,30),
> OA(22),OAL (22),RAY1 (4),\operatorname{IFM}(7,2,3),\operatorname{FBW}(7,2,3)
> (C-98),D(98),E(98),STV(2),OLDTFW(2),IOLDFW(2)
EQUIVALENCE (C,S(1)),(D,S(99)),(E,S(197))
COMMON /CVAERD/ITIME,THET,ICH11,ICH12,ISQ11,ISQ12,ITIMEL,THETL,
>>
> ,IEND4,IX,ISVIF,OA,OAL

```
    COMMON /CRAYL/ICH, NSAT, IGZC, STV, DC, RAY1, RAY2,ICE, ISKIP
    DATA OLDTPM/0.,0.1.OLDTFW/0.,0./, IOLDFW/0,0/
    DATA C1/. \(1544, .175, .1658, .1677, .1757, .1759 /\)
    DATA \(\mathrm{C} 2 / 4.5,4.6,19.02,11.85,11.57,-1.827 /\)
DATA C/ 20.0, 19.6, 19.4, 19.4, 19.9, 21.8, 24.4.
\(\begin{array}{ll}2 & 101.5,79.3,66.8,57.3,51.1,45.9,44.2 \\ 3 & 12.2,11.5,11.1,11.0,11.1,12.0,13.5,\end{array}\)
\(4 \quad 23.6,20.7,19.2,18.6,18.4,18.6,19.6\),
\(5 \quad 35.2,35.3,35.8,36.7,37.9,39.4,41.4\),
\(6 \quad 0.0, \quad 0.0, \quad 0.0, \quad 0.0,10.0, \quad 0.0, \quad 0.0\),
\(7 \quad 10.7,10.4,10.6,10.9,11.4,12.1,12.8\),
\(10.30,0.27,0.25,0.24,0.24,0.24,0.25\),
\(20.00,0.00,0.00,0.00,0.00,0.00,0.00\),
3 1.45, 1.13, \(0.95,0.82,0.73,0.65,0.63\),
\(40.15,0.15,0.15,0.16,0.17,0.17,0.19\),
\(50.24,0.24,0.23,0.23,0.24,0.26,0.29\),
\(7 \quad 0.61,0.61,0.62,0.64,0.66,0.68,0.72\) /
    DATA D/ 80.0, 72.1, 68.5, 72.1, 77.8, 87.3.106.6,
    \(2 \quad 7.6,7.6,7.9,8.4,9.6,12.2,16.3\),
    \(3 \quad 26.0,24.2,23.3,23.5,24.4,26.2,29.2\),
        \(9.7,8.5,8.2,8.1,8.1,8.1,8.3\),
        12. \(2,11.3,11.1,11.1,11.1,11.2,11.6\),
        \(\begin{array}{llllll}12.2, & 11.3, & 11.1, & 11.1, & 1.0, & 0.0, \\ 0.0 & 0.0, & 0.0, & 0.0, & 0.0, & 0.0\end{array}\)
        \(22.2,17.6,16.7,16.3,16.4,19.7,20.4\),
        \(\begin{array}{llllll}22.2, & 17.12, & 0.11, & 0.11, & 0.11, & 0.11, \\ 0.13, & 0.11,\end{array}\)
        \(0.00,0.00,0.00,0.00,0.00,0.00,0.00\),
        \(0.10,0.10,0.10,0.11,0.12,0.16,0.21\),
        \(0.24,0.19,0.18,0.18,0.18,0.21,0.22\),
        \(0.76,0.69,0.65,0.69,0.74,0.83,1.02\),
        \(0.37,0.34,0.33,0.33,0.34,0.37,0.41\),
```

7 0.14, 0.13, 0.13, 0.13, 0.13, 0.13,0.13/
DATA E/124.4, 88.8, 73.7, 61.7, 53.9, 47.2, 44.2,
11.0, 10.3, 10.1, 10.1, 10.4, 11.7, 14.1,
42.9, 42.9, 42.8, 42.9, 42.9, 42.9, 42.9,
22.8, 21.2, 20.4, 20.7, 21.3, 22.8, 29.3,
12.7, 12.7, 13.2, 15.8, 21.8, 29.3, 42.4,
0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0,
19.8, 19.6, 20.4, 22.5, 24.2, 46.2, 72.8,
0.26, 0.24, 0.23, 0.23, 0.24, 0.26, 0.33,
0.00, 0.00, 0.00, 0.00, 0.00, 0.00, 0.00,
0.15,}00.14, 0.14, 0.14, 0.14, 0.16, 0.19
0.23, 0.23, 0.24, 0.26, 0.29, 0.54, 0.86,
1.27, 0.91, 0.76, 0.63, 0.55, 0.48, 0.28,
0.60, 0.60, 0.60, 0.60, 0.60, 0.60, 0.60,
0.16, 0.16, 0.16, 0.19, 0.27, 0.36, 0.52/
DATA IFM/ 55, 73, 52, 63, 33, 0, 42,
> 63, 0, 73, 42, 55, 52, 33,
> 73, 52, 42, 48, 55, 0, 63,
> 48, 0, 52, 63, 73, 42, 55,
> 73, 52, 28,65,55, 0, 0,63,
> 65, 0, 52, 63, 73, 28, 55 /
DATA FBW/ 29.8, 20.0, 22.5, 22.7, 0.0, 0.0, 23.0,
> 22.7, 0.0, 20.0, 23.0, 29.8, 22.5, 0.0,
> 29.0, 21.0, 19.0, 0.0, 25.0, 0.0, 20.5,
0.0, 0.0, 21.0, 20.5, 29.0, 19.0, 25.0,
15.8, 20.1, 0.0, 0.0, 19.5, 0.0, 21.0,
0.0, 0.0, 20.1, 21.0, 15.8, 0.0, 19.5//
ICE=0
IF(IFW.LT.1.OR.IFW.GT.7)GOTO 300
IF (TPM (ICH).EQ.OLDTPM (ICH)) GOTO 100
IF (TPM (ICH).GT.40.0.OR.TPM (ICH).LT. -20.0)TPM (ICH)=OLDTPM (ICH)
OLDTPM (ICH) =TPM (ICH)
DC (ICH) = EXP (C1 (ICH,NSAT)*(TPM (ICH)-C2 (ICH,NSAT)))
100 IF (TFW.EQ.OLDTFW(ICH).AND.IFW.EQ.IOLDFW(ICH)) GOTO 200
IF (TFW.GT.40.0.OR.TFW.LT. -20.0)TFW=OLDTFW (ICH)
OLDTFW (ICH) =TFW
IOLDFW (ICH) = IFW
S1=S (IFIX(TFW+30.)/10,IFW,ICH,NSAT)
S2=S(IFIX(TFW+40.)/10,IFW,ICH,NSAT)
STV (ICH) =S1+(TFW/10.-IFIX (TFW)/10)*(S2-S1)
200 IF(ICH.EQ.2)GOTO 250
RAY1 (1) = (ICH11-DC(1))*STV (1)
RAY1 (2) = (ICH12-DC(1))*STV(1)
RAY1 (3) =(ICH11L-DC (1))*STV (1)
RAY1 (4) =(ICH12L-DC (1))*STV (1)
GOTO 500
250 RAY2 = (ICH2-DC (2))*STV (2)
GOTO 500
RAY2=0
DO 400 I=1,4
RAY1 (I) =0.
RETURN
IF(IGZC.EQ.0)RETURN

```
\(C\) only open and read the data file the first time RAYL is called.
```

IF(ISKIP.EQ.0) CALL GZREAD (GAL, ZOD)
IF(ICH.EQ.2)GOTO }60
LAM=IFM(IFW,ICH,NSAT)
IF(LAM.NE.0)GOTO 530
ICE=4
RETURN
CONTINUE
ICE $=1$
IF (OA (13) .LT. -180..OR.OA(13). GT . 180.) ICE $=2$
IF (OA (14). LT. -90..OR.OA (14).GT.90.) ICE=2
IF (OA (15). LT . -90..OR.OA (15).GT.90.) ICE=2
IF (OA (16). LT . -180 . OR.OA (16).GT. 180.) ICE=2
IF (ICE.EQ.2)RETURN
$\operatorname{IF}(\mathrm{OA}(13) . \operatorname{LT} \cdot 0) .\mathrm{OA}(13)=O A(13)+360$.
$\operatorname{IF}(O A L(13) . L T \cdot 0) O A L.(13)=O A L(13)+360$.
IF (OAL (13).GT. 350..AND.OA (13).LT. 10.) OA (13) $=O A(13)+360$
IF (OAL (13) . LT . 10..AND.OA (13). GT . 350.) OAL (13) $=\mathrm{OAL}(13)+360$.
DRA=OAL (13)-OA (13)
DDEC=OAL (14)-OA (14)
DELAT=OAL (15) -OA (15)
OAL (16) =ABS (OAL (16))
$\mathrm{OA}(16)=\mathrm{ABS}(\mathrm{OA}(16))$
DELON=OAL (16) -OA (16)
$C$ * the following is changed in this program
1/7/91
$\operatorname{RA}(1)=\operatorname{IFIX}(\mathrm{OA}(13)+0.25 * \operatorname{DRA}) / 3+1$
$\operatorname{RA}(2)=\operatorname{IFIX}\left(\mathrm{OA}(13)+0.25 * \mathrm{DRA}^{*} 3.\right) / 3+1$
$\operatorname{RA}(3)=\operatorname{IFIX}(\operatorname{OAL}(13)+0.25 * \operatorname{DRA}) / 3+1$
$\operatorname{RA}(4)=\operatorname{IFIX}($ OAL $(13)+0.25 *$ DRA* 3.$) / 3+1$
$\operatorname{DEC}(1)=\operatorname{IFIX}(\mathrm{OA}(14)+0.25 * \mathrm{Ddec}+90) / 3+$.
$\operatorname{DEC}(2)=\operatorname{IFIX}\left(\mathrm{OA}(14)+0.25 * \mathrm{Ddec}^{*} 3+90.\right) / 3+1$
$\operatorname{DEC}(3)=\operatorname{IFIX}($ OAL $(14)+0.25 * \operatorname{Ddec}+90) / 3+$.
$\operatorname{DEC}(4)=\operatorname{IFIX}(\operatorname{OAL}(14)+0.25 * \operatorname{Ddec} * 3 .+90) / 3+$.
$\operatorname{ELAT}(1)=\operatorname{IFIX}(\operatorname{ABS}(O A(15)+0.25 * \operatorname{DELAT})) / 3+1$
$\operatorname{ELAT}(2)=\operatorname{IFIX}(\operatorname{ABS}(\mathrm{OA}(15)+0.25 * \operatorname{DELAT} * 3)) / 3+1$
$\operatorname{ELAT}(3)=\operatorname{IFIX}(\operatorname{ABS}(\operatorname{OAL}(15)+0.25 *$ DELAT $)) / 3+1$
$\operatorname{ELAT}(4)=\operatorname{IFIX}\left(\operatorname{ABS}\left(\operatorname{OAL}(15)+0.25 * \operatorname{DELAT}^{*} 3\right)\right) / 3+1$
$\operatorname{ELON}(1)=\operatorname{IFIX}(O A(16)+0.25 *$ DELON $) / 3+1$
$\operatorname{ELON}(2)=\operatorname{IFIX}(\mathrm{OA}(16)+0.25 *$ DELON* 3.$) / 3+1$
$\operatorname{ELON}(3)=\operatorname{IFIX}($ OAL $(16)+0.25 *$ DELON $) / 3+1$
$\operatorname{ELON}(4)=\operatorname{IFIX}(O A L(16)+0.25 *$ DELON* 3.$) / 3+1$
IF (LAM.EQ.42.OR.LAM.EQ.33.OR.LAM.EQ.28) LI = 1
IF (LAM.EQ.52.OR.LAM.EQ.48) LI $=2$
IF (LAM.EQ.55) LI $=3$
IF (LAM.EQ.63.OR.LAM.EQ.65) $\quad$ LI $=4$
$I F(L A M . E Q .73) \quad L I=5$
DO $550 \mathrm{I}=1,4$
IF (RA (I).GT. 120.OR.RA (I).LT.1)RA (I) =1
$\operatorname{IF}(\operatorname{ELON}(I) . G T \cdot 60) \operatorname{ELON}(I)=60$
IF (ELON (I).LT. 1) ELON (I) =1
$\operatorname{IF}(\operatorname{ELAT}(I) . \operatorname{GT} \cdot 30) \operatorname{ELAT}(I)=30$
$\operatorname{IF}(\operatorname{DEC}(I) . G T \cdot 60) \operatorname{DEC}(I)=60$
$C$ * following elon(I) \& elat(I) replace faulty RA(I) \& DEC(I) IF (GAL (LI, RA (I) , DEC (I)) .GT. O.OR.
$>$ ZOD(LI,elon(I), elat(I)).GT.0.) THEN
GOTO 545
ENDIF

```
\(\operatorname{GAL}(L I, R A(I), \operatorname{DEC}(I))=0\).
\(\operatorname{ZOD}(L I, R A(I), \operatorname{DEC}(I))=0\).
ICE=-1
C * original "DEC(I))*16." in 545 changed
\(C\) * original "IL" in cont 545 changed to "li"

CONTINUE
RETURN
600 ICE=1 19 .LT. -180..OR.OAL (19).GT. 180.) ICE \(=2\)

IF (OAL (21).LTT. -90..OR.OAL (21).GT.90.) ICE=2
IF (OAL (22) .LT . -180..OR.OAL (22).GT . 180.) ICE=2
IF (ICE.EQ.2)RETURN
\(\operatorname{IF}(\) OAL (19) . LT \(\cdot 0\).\() OAL (19) =\) OAL (19) +360 .
IRA2 \(=\operatorname{IFIX}(\) OAL (19) ) \(/ 3+1\)
IF (IRA2.GT.120) IRA \(2=120\)
\(\operatorname{IDEC} 2=\operatorname{IFIX}(\operatorname{OAL}(20)+90) / 3+\).
IF (IDEC2.GT.60) IDEC2 \(=60\)
\(\operatorname{IELAT} 2=\operatorname{IFIX}(\operatorname{ABS}(\operatorname{OAL}(21))) / 3+1\)
IF (IELAT2.GT.30) IELAT2 \(=30\)
\(\operatorname{IELON} 2=\operatorname{IFIX}(\operatorname{ABS}(\operatorname{OAL}(22))) / 3+1\)
IF (IELON2.GT . 60) IELON2 \(=60\)
LAM \(=\) IFM (IFW, ICH, NSAT) \(\quad\) LI \(=1\)
IF (LAM.EQ.42.OR. LAM.EQ. 33 . OR. LAM \(=2\)
\(L T=3\)
IF (LAM.EQ.63.OR.LAM.EQ.65) LI \(=4\)
\(\operatorname{IF}(L A M . E Q .73) \quad \mathrm{LI}=5\)
IF (LAM.NE.0) GOTO 640
ICE=4
RETURN
IF (GAL (LI, IRA2, IDEC2).GT.0.) GOTO 650
GAL (LI, IRA2, IDEC2) \(=0\).
ICE=-1
C * following "li" replaces original faulty IL
650 RAY2 = RAY2 - (GAL (LI, IRA2, IDEC2) +
\& ZOD(1i, IELON2, IELAT2))*FBW (IFW, ICH, NSAT)
RETURN
900 ICE \(=3\)
RETURN
END
SUBROUTINE GZREAD (GAL, ZOD)
C Subroutine GZREAD returns a \(120 \times 60\) array of galaxy data and a \(60 \times 30\)
\(C\) array of zodiacal light data for each of 5 wavelengths.
DIMENSION \(\operatorname{GZDAT}(45000), \operatorname{GAL}(5,120,60), \operatorname{ZOD}(5,60,30)\)
OPEN (12,FILE='SPRLC\$DISK1:[VAECOMMON.IND]GALZOD.DAT', TYPE='OLD',
\& READONLY)
\(\operatorname{READ}(12,1000, \mathrm{END}=5) \mathrm{GZDAT}\)
1000 FORMAT (1X, 15F5.2)
\(5 \quad\) IC \(=0\) are 5 pairs of galaxy and zodiacal light data (one for each
\(C\) wavelength).
wavelength).
C Read galaxy data for one wavelength
DO \(20 \mathrm{~K}=1,60\)
DO \(30 \mathrm{~J}=1,120\)
        \(I C=I C+1\)
        \(\operatorname{GAL}(I, J, K)=\operatorname{GZDAT}(I C)\)
CONTINUE
CONTINUE
Read zodiacal light data for one wavelength
DO \(40 \mathrm{~N}=1\), 30
DO \(50 \mathrm{M}=1,60\)
\(I C=I C+1\)
\(\operatorname{ZOD}(\mathrm{I}, \mathrm{M}, \mathrm{N})=\operatorname{GZDAT}(\mathrm{IC})\)
CONTINUE
CONTINUE
CONTINUE
RETURN
END
SUBROUTINE MAXMUL1 ( \(\mathrm{A}, \mathrm{B}, \mathrm{C}\) )
C IMPLICIT DOUBLE PRECISION ( \(\mathrm{A}-\mathrm{H}, \mathrm{O}-\mathrm{Z}\) )
DIMENSION A \((3,3), B(3), C(3)\)
\(C(1)=A(1,1) * B(1)+A(1,2) * B(2)+A(1,3) * B(3)\)
\(\mathrm{C}(2)=\mathrm{A}(2,1) * \mathrm{~B}(1)+\mathrm{A}(2,2) * \mathrm{~B}(2)+\mathrm{A}(2,3) * \mathrm{~B}(3)\)
\(\mathrm{C}(3)=\mathrm{A}(3,1) * \mathrm{~B}(1)+\mathrm{A}(3,2) * \mathrm{~B}(2)+\mathrm{A}(3,3) * \mathrm{~B}(3)\)
RETURN
END

\section*{Appendix 3}

The following is a listing of the Pascal (VAX-11) program used to invert the AE-E Visible Airglow Experiment measurements of the \(\mathrm{O}^{+}(2 \mathrm{D}-2 \mathrm{P})\) twilight airglow emission at \(7320 \AA\).
```

program TWIFITTER (input,output);
program for inverting multi-directional twilight
AE VAE 7320 A observations developed for the project: }
An Assessment of Twilight Airglow Inversion Procedures,
Using Atmosphere Explorer Observations }
by I.C.McDade }
under NASA Grant NAG 5-1502
this self-contained VAX-11 PASCAL program will return:
the atomic oxygen scale height, H (km)
the O+(2}\textrm{P})\mathrm{ ionization frequency, Iinf ( }\mp@subsup{\textrm{sec}}{}{-1}
and the O-atom density \& 250 km, 0250 (cm-3)
that best fit the input 7320-30 A column emission rates
many of the functions and procedures are taken from:
Numerical Recipes: The Art of Scientific Computing
by W.H.Press, B.P.Flannery, A.Teukolsky and W.T.Vetterling
Cambridge University Press, New York, 1986
other aspects are based on the formulation described by

```
```

function gammln (xx: real): real;

```
function gammln (xx: real): real;
    const
    const
        stp \(=2.50662827465\);
        stp \(=2.50662827465\);
        half \(=0.5\);
        half \(=0.5\);
        one \(=1.0\);
        one \(=1.0\);
        fpf \(=5.5\);
        fpf \(=5.5\);
    var
    var
        x,tmp,ser: double;
        x,tmp,ser: double;
        j: integer;
        j: integer;
        cof: array[1..6] of double;
        cof: array[1..6] of double;
begin
begin
    \(\operatorname{cof}[1]:=76.18009173\);
    \(\operatorname{cof}[1]:=76.18009173\);
    \(\operatorname{cof}[2]:=-86.50532033\);
    \(\operatorname{cof}[2]:=-86.50532033\);
    \(\operatorname{cof}[3]:=24.01409822\);
    \(\operatorname{cof}[3]:=24.01409822\);
    \(\operatorname{cof}[4]:=-1.231739516\);
    \(\operatorname{cof}[4]:=-1.231739516\);
    \(\operatorname{cof}[5]:=0.120858003 \mathrm{e}-2\);
    \(\operatorname{cof}[5]:=0.120858003 \mathrm{e}-2\);
    \(\operatorname{cof}[6]:=-0.536382 e-5\);
    \(\operatorname{cof}[6]:=-0.536382 e-5\);
    \(x:=x X\)-one;
    \(x:=x X\)-one;
    tmp: \(=x+f p f ;\)
```

    tmp: \(=x+f p f ;\)
    ```
        \(\}\)
```

tmp:= (x+half)*ln(tmp)-tmp;
ser:= one;
for j:= 1 to 6 do
begin
x:= x+one;
ser:= ser+cof[j]/x
end;
gammln:= sngl(tmp+ln(stp*ser))
end;
procedure gser (a,x: real; var gamser,gln: real);
label 1
const
itmax = 100;
eps = 3.0e-7;
var
n: integer
sum,del,ap: real;
begin
gln:= gammln(a);
If ( }x<=0.0) the
begin
if (x < 0.0) then
begin
writeln('pause in GSER - x less than 0');
readln
end;
gamser:= 0.0
end
else
begin
ap:= a;
sum:= 1.0/a;
del:= sum;
for n:= 1 to itmax do
begin
ap:= ap+1.0;
del:= del*x/ap;
sum:= sum+del;
if (abs(del) < abs(sum)*eps) then
goto 1
end;
readln;
1:
gamser:= sum*exp (-x+a* ln (x)-gln)
end
end;
procedure gcf {a,X: real; var gammcf,gln: real};
label
1;
const
itmax = 100;
eps = 3.0e-7;
var
n: integer;
gold,g,fac,b1,b0,anf,ana,an,al,a0: real;
begin
gln:= gammln(a);
gold:= 0.0;

```
```

a0:= 1.0;
a1:= x;
b0:= 0.0;
b1:= 1.0;
fac:= 1.0;
for n:= 1 to itmax do
begin
an:= 1.0*n;
ana:= an-a;
a0:= (a1+a0*ana)*fac;
b0:= (b1+b0*ana)*fac;
anf:= an*fac;
al:= x*a0+anf*a1;
b1:= x*b0+anf*b1;
if (a1 <> 0.0) then
begin
fac:= 1.0/a1;
g:= b1*fac;
if (abs((g-gold)/g) < eps) then
goto 1;
gold:= g
end
end;
writeln('pause in GCF-a too large,itmax too small');
readln;
1:
gammcf:= exp (-x+a* ln (x)-gln)*g
end;
function gammp (a,x: real): real;
var
gammcf,gln: real;
begin
if ((x<0.0) or (a <= 0.0)) then
begin
writeln('pause in GAMMP-invalid arguments');
readln
end;
if (x < (a+1.0)) then
begin
gser (a,x,gammc f,gln);
gammp:= gammcf
end
elge
begin
gcf(a,x,gammcf,gln);
gammp:= 1.0-gammcf
end
end;
function erf (x: real): real;
begin
if (x<0.0) then
begin
erf:= - gammp (0.5,sqr (x))
end
else
begin
erf:= gammp (0.5,\operatorname{sqr}(x))
end
end;

```
```

const
TWO_PI = 6.283185307179586476925287; {value of Two pi}
PI = 3.141592653589793238462643; {Value of PI}
R = 6370.0; {Approximate Earth Radius}

```
[GLOBAL]
function INRANGE (VALUE,MIN,MAX: REAL): REAL;
    var
        NRANGES: REAL;
        N: INTEGER;
begin (VALUE < MIN) or (VALUE >= MAX) then \{value outside range\}
    if (VALUE < MIN) or (VALUE \{get how many times\}
        begin \(\quad \begin{aligned} & \text { \{get how many times } \\ & \text { \{convert from min to max to\} }\end{aligned}\)
            VALUE:= VALUE-MIN; \(\quad\{0\) to (max-min) \(\}\)
MAX: \(=\) MAX-MIN;
            NRANGES:= VALUE/MAX; \{by dividing by range\}
            \(\mathrm{N}:=\) TRUNC (NRANGES) ;
            if NRANGES < 0.0 then
                \(\mathrm{N}:=\mathrm{N}-1\);
            \(\begin{array}{cl}\mathrm{N}:=\mathrm{N}-1 ; & \text { \{make trunc } \\ \text { VALUE: }=\mathrm{MAX} \text { (NRANGES-N); } & \text { \{get the in range value \}}\end{array}\)
        \{make trunc work correctly\}
            INRANGE: = VALUE+MIN;
        \{add on the minimum offset \}
        end
    else \{otherwise the value\}
        begin \(\quad\) \{is already in range \(\}\)
            INRANGE: = VALUE; \(\quad\) \{so do nothing\}
        end: \(\begin{cases}\text { \{back to caller\} }\end{cases}\)
    end;
    [EXTERNAL]
    function MTHSASIN(X: REAL): REAL;
    EXTERNAL;
    [EXTERNAL]
    function MTH\$ACOS (X: REAL) : REAL;
    EXTERNAL;
    [GLOBAL]
    function ASIND(X: REAL): REAL;
    begin ASIND: \(=360.0 *\) INRANGE ( MTH\$ASIN \(\left.(X), 0, T W O \_P I\right) / T W O \_P I ;\)
    end;
    [GLOBAL]
    function SIND(X: real): real;
    begin
        SIND:= SIN(X*TWO_PI/360.0);
    end;
    [GLOBAL]
    function \(\operatorname{CoSD}(\mathrm{X}:\) real): real;
    begin
        \(\operatorname{CosD}:=\cos \left(X * T W O \_P I / 360.0\right) ;\)
    end;
    function angle (v11,v12,v13,v21,v22,v23: real): real;
    \{returns angle in degrees between two Cartesian vectors\}
        var
            \(a, b, c: r e a l ;\)
    begin
        \(\mathrm{a}:=\mathrm{v} 11 * \mathrm{v} 21+\mathrm{v} 12^{*} \mathrm{v} 22+\mathrm{v} 13^{*} \mathrm{v} 23\);
        \(\mathrm{b}:=\mathrm{v} 11^{*} \mathrm{~V} 11+\mathrm{v} 12^{*} \mathrm{v} 12+\mathrm{v} 13^{*} \mathrm{~V} 13\);
```

    c:= v21*v21+v22*v22+v23*v23;
    angle:= MTH$ACOS (a/sqrt (b*c))*360.0/TWO_PI;
    end;
START OR MAIN
type
glndata = array[1..1000] of real;
glmma = array[1..3] of real;
glncabynca = array[1..3,1..3] of real;
gllista = array[1..3] of integer
glnalbynal = array[1..3,1..3] of real;
glcovar = array[1..3,1..3] of real;
glnpbynp = array[1..3,1..3] of real
glnpbymp = array[1..3,1..3] of real;
glnp = array[1..3] of integer;
vect = array[1..3,1..1000] of real; {for direction vectors}
var pv,vv,sv: vect; {satellite position,line-of-sight \& sun GEI vectors}
XV,Y SIG, alt,za,fit,time,sza: glndata; {time \& X should be same}
A,DYDA: glmma;
LISTA: gllista;
covar,ALFA,dummy: glncabynca;
CHISQ,ALAMDA,chisqo,deltachisq,ln0250,0500: real;
Xs,XsN2,Flux,WXs: array[1..23] of real;
N2250,KO,KN2: real;
I,J,ndata,mfit,maxhindx,deltah: integer;
infile,vecfile,outfile: text;
infilename,vecfilename,outfilename: varying[80] of char;
glochisq: real;
glbeta: glmma; {for MRQMIN}
{---------------------------BRIGHT-----------------------------------------------
procedure BRIGHT (I: integer; H,Iinf,lnO250: real; var INT: real);
Var,V,0250,Vtemp,Ohm,N2hm: real;
h1,h2,za1,za2,hm, chim: real;
a,X,Y,F,TAU,XN2,YN2,HN2,FN2,TAUN2: real;
k,j: integer;
IFXWXS: real;
pos1,posm,sun: array[1..3] of real;
B: real.
begin
0250:= exp(lnO250);
HN2:= H*16.0/28.0;
IFXWXs:= 6.7857e-18; {Integrated flux shape x xsection}
Int:= 0.0;
h1:= alt[I];
zal:= za[I];
pos1[1]:= pv[1,I];
pos1[2]:= pv[2,I];
pos1[3]:= pv[3,I];
sun[1]:= sv[1,I];
sun[2]:= sv[2,I];
sun[3]:= sv[3,I];
for K:= 1 to maxhindx do
begin {integration along line of sight I}
h2:= h1+deltah;
B:= za1-za2;

```
```

W:= SIND(B)* (R+h1)/SIND(za2);
if (B = 0.0) then
W:= h2-h1;
posm[1]:= pos1[1]+0.5*W*Vv[1,I]; {GEI vectors for mid element}
posm[2]:= pos1[2]+0.5*W*VV[2,I];
posm[3]:= pos1[3]+0.5*W*vv[3,I];
chim:= ANGLE(sun[1],sun[2],sun[3], posm[1],posm[2],posm[3]);
hm:= (h1+h2)/2; {mean altitude of element}
if (chim >= 90.0) then {Chapman Funcs from eql7.21 of}
{Banks\&Kockarts}
begin
a:= (R+hm)*COSD (chim-90.0)-R; {minimum ray height}
if (a > 100.0) then
begin
Y:= (R+a)/H; {Y for O}
YN2:= (R+a)/HN2; {Y for N2}
F:= sqrt (pi*Y/2.0)*(1+erf(sqrt(Y/2.0)*COSD(chim)/SIND (chim)));
FN2:=sqrt (pi*YN2/2.0)*(1+erf(sqrt (YN2/2.0)*COSD(chim)/SIND(chim)));
Ohm:= 0250* exp (-(hm-250)/H); {[O] at hm}
N2hm:= N2250* exp(-(hm-250)/HN2); {[N2] at hm}
V:= 0.0;
for j:= 1 to 23 do {start of wavelength loop}
begin
TAU:= 0250* exp (- (a-250)/H)*H*1e5*Xs[j]*F;
TAUN2:= N2250* exp (-(a-250)/HN2)*HN2*1e5*XsN2[j]*FN2;
Vtemp:= Ohm*exp(-TAU)*exp(-TAUN2)*Flux[j]*WXs[j];
Vtemp:= Vtemp*Iinf/IFxWXs; {normalize to parameter Iinf}
Vtemp:= Vtemp*0.781*0.219/(0.219+KO*Ohm+KN2*N2hm);
V:= V+Vtemp;
end; {end of wavelength loop}
INT:= INT+0.1*W*V;
end; {end of if a>100.0}
end; {end of if chim >=90}
if (chim < 90.0) then {Chapman Funcs from eq17.17 of }
{Banks \& Kockarts}
begin
X:= (R+hm)/H;
XN2:= (R+hm)/HN2;
F:=sqrt(pi*X/2.0)* exp((X/2)*sqr (COSD(chim)));
F:=F* (1-erf(sqrt(X/2.0)*COSD(chim)));
FN2:= sqrt (pi*XN2/2.0)*exp ((XN2/2)*sqr (COSD (chim)));
FN2:= FN2*(1-erf(sqrt(XN2/2.0)*COSD(chim)));
Ohm:= 0250* exp (- (hm-250)/H); {[0] at hm}
N2hm:= N2250* exp (-(hm-250)/HN2); {[N2] at hm}
V:= 0.0;
for j:= 1 to 23 do { start of wavelength loop}
begin
TAU:= Ohm*H*1e5*Xs[j]*F; {attenuation due to 0}
TAUN2:= N2hm*HN2*1e5*XsN2[j]*FN2; {"" due to N2}
Vtemp:= Ohm*exp (-TAU)*exp(-TAUN2)*Flux[j]*WXs[j]; {prod at hm}
Vtemp:= Vtemp*Iinf/IFxWXs; {normalize to parameter Iinf}
Vtemp:= Vtemp*0.781*0.219/(0.219+KO*Ohm+KN2*N2hm);
V:= V+Vtemp;
end; {end of wavelength loop}
INT:= INT+0.1*W*V; {add contribution from chim,hm and }
{convert to Rayleighs)
end; {end of if chim <90}
fit[I]:= INT; {keep fit to obs I for output}
h1:= h2;
za1:= za2;
pos1[1]:= pos1[1]+W*VV[1,I]; {GEI vectors next element}
pos1[2]:= pos1[2]+W*Vv[2,I];
pos1[3]:= pos1[3]+W*vv[3,I];

```
end; \{end of integration along line of sight \(I\}\)
end; \{of procedure BRIGHT\}

procedure FUNCS (i: integer; a: glmma; var \(y\) : real; var dyda: glmma); var ai,af,yi,yf: real;
begin BRIGHT (I, A[1], A[2], A[3], Y) ; ai:= 0.999*A[1]; af:=1.001*A[1]; BRIGHT (I, ai, A[2],A[3],yi); BRIGHT (I, af, A[2],A[3],yf); DYDA[1]: = (yi-yf)/(ai-af); ai: \(=0.999^{* A}\) [2]; af: \(=1.001 * A[2] ;\)
BRIGHT (I, A[1], ai, A[3],Yi); BRIGHT (I,A[1],af,A[3],yf); DYDA[2]:= (yi-yf)/(ai-af); ai \(:=0.999 * A[3]\); af: \(=1.001 * A[3]\); BRIGHT (I, A[1], A[2], ai, yi); BRIGHT (I, A[1], A[2], af,yf); DYDA[3]: = (Yi-yf)/(ai-af);
end; \{end of procedure FUNCS\}
```

{--------------------------GAUSSJ

```
    procedure gaussj (var a: glnpbynp; \(n, n p: i n t e g e r ; ~ v a r ~ b: ~ g l n p b y m p ; ~\)
m, mp: integer);
    var
        big, dum, pivinv: real;
        i,icol,irow,j,k,1,11: integer;
        indxc,indxr,ipiv: glnp;
    begin
    for j:= 1 to \(n\) do
        begin
            ipiv[j]:= 0
            end;
        for \(i:=1\) to \(n\) do
        begin
            big:= 0.0;
            for j:= 1 to \(n\) do
                begin
                    if (ipiv[j] <> 1) then
                        begin
                                for \(k:=1\) to \(n\) do
                                begin
                        if (ipiv[k] = 0) then
                        begin
                            if (abs \((a[j, k])>=b i g)\) then
                        begin
                                big:= abs \((a[j, k])\);
                                irow:= j;
                        icol:= k
                                end
                            end
                        else if (ipiv[k] > 1) then
                        begin
                        writeln('pause 1 in gaussj-singular matrix');
                                readln
                        end
```

            end
        end
    end;
    ipiv[icol]:= ipiv[icol]+1;
if (irow <> icol) then
begin
for l:= 1 to n do
begin
dum:= a[irow,l];
a[irow,1]:= a[icol,l];
a[icol,l]:= dum
end;
for 1:= 1 to m do
begin
dum:= b[irow,1];
b[irow,l]:= b[icol,l];
b[icol,l]:= dum
end
end;
indxr[i]:= irow;
indxc[i]:= icol;
if (a[icol,icol] = 0.0) then
begin
writeln('pause 2 in gaussj-singular matrix');
readln
end;
pivinv:= 1.0/a[icol,icol];
a[icol,icol]:= 1.0;
for 1:= 1 to n do
begin
a[icol,1]:= a[icol,1]*pivinv
end;
for 1:= 1 to m do
begin
b[icol,1]:= b[icol,1]*pivinv
end;
for ll:= 1 to n do
begin
if (1l <> icol) then
begin
dum:= a[11,icol];
a[11,icol]:= 0.0;
for 1:= 1 to n do
begin
a[11,1]:= a[11,1]-a[icol,1]*dum
end;
for l:= 1 to m do
begin}11:= b[11,1]-b[icol,1]*dum
end
end
end
end;
for 1:= n downto 1 do
begin
if (indxr[1] <> indxc[1]) then
begin
for k:= 1 to n do
begin
dum:= a[k,indxr[1]];
a[k,indxr[l]]:= a[k,indxc[1]];
a[k,indxc[11]]:= dum

```

\section*{end \\ end \\ end}
end; \{end of procedure gaussj\}
\{----------------------------COVSRT-
procedure covsrt (var covar: glcovar; ncvm: integer; ma: integer; lista: gllista; mfit: integer);

\section*{var}
j,i: integer;
swap: real;
begin
for j:= 1 to ma-1 do
begin
for \(1:=j+1\) to ma do
begin
covar[i,j]:= 0.0
end
end;
for i:= 1 to mfit-1 do
begin
for \(j:=i+1\) to mfit do
begin
If (1ista[j] > lista[i]) then
begin covar[1ista[j], Iista[i]]:= covar[i,j]
end
else
begin
covar[lista[i],lista[j]]:= covar[i,j]
end
end
end;
swap:= covar[1,1];
for \(j:=1\) to ma do
begin
covar \([1, j]:=\operatorname{covar}[j, j] ;\)
covar[j,j]:= 0.0
end;
covar[lista[1], lista[1]]:= swap;
for \(j:=2\) to mfit do
begin
covar[lista[j],lista[j]]:= covar[1,j]
end;
for \(j:=2\) to ma do
begin
for \(i:=1\) to \(j-1\) do
begin
covar[i,j]:= covar[j,i]
end
end
end; \{end of procedure covstr\}

procedure mrqcof ( \(x, y, s i g:\) glndata; ndata: integer; var a: glmma; mma: integer; lista: gllista; mfit: integer; var alpha: glnalbynal; var beta: glmma; nalp: integer; var chisq: real);
```

            var
            k,j,i: integer;
    ```
```

    ymod,wt,sig2i,dy: real;
    dyda: glmma;
    begin
for j:= 1 to mfit do
begin
for k:= 1 to j do
begin
alpha[j,k]:= 0.0
end;
beta[j]:= 0.0
end;
chisq:= 0.0;
for i:= 1 to ndata do
begin
FUNCS(i,a,ymod,dyda):
sig2i:= 1/(sig[i]*sig[i]);
dy:= y[i]-ymod;
for j:= 1 to mfit do
begin
wt:= dyda[lista[j]]*sig2i;
for k:= 1 to j do
begin
end;
beta[j]:= beta[j]+dy*Wt
end;
chisq:= chisq+dy*dy*sig2i
end;
for j:= 2 to mfit do
begin
for k:= 1 to j-1 do
begin
alpha[k,j]:= alpha[j,k]
end
end
end; {end of procedure MRQCOF}
----------------------MRQMIN-----------------------------------------
procedure mrqmin (x,y,sig: glndata; ndata: integer; var a: glmma;
mma: integer; lista: gllista; mfit: integer; var covar,alpha:
glncabynca; nca: integer; var chisq,alamda: real);
label 99;
var
k,kk,j,ihit: integer;
atry,da: glmma;
oneda: glncabynca;
begin
if (alamda < 0.0) then
begin
kk:= mfit+1;
for j:= 1 to mma do
begin
ihit:= 0;
for k:= 1 to mfit do
begin
if (lista[k] = j) then
init:= ihit+1
end;
if (init = 0) then
begin
lista[kk]:= j;

```
```

                kk:= kk+1
            end
            else if (ihit > 1) then
                begin
                writeln('pause 1 in routine MRQMIN');
                writeln('Improper permtuation in LISTA');
                readln;
            end
        end;
    if (kk <> (mma+1)) then
        begin
            writeln('pause 2 in routine MRQMIN');
            writeln('Improper permtuation in LISTA');
            readln;
        end;
    alamda:= 0.001;
    MRQCOF(x,y,sig,ndata,a,mma,lista,mfit,alpha,glbeta,nca,chisq);
    glochisq:= chisq;
    for j:= 1 to mma do
        begin
            atry[j]:= a[j]
        end
    end;
    for j:= 1 to mfit do
begin
for k:= 1 to mfit do
begin
covar[j,k]:= alpha[j,k]
end;
covar[j,j]:= alpha[j,j]*(1.0+alamda);
oneda[j,1]:= glbeta[j]
end;
GAUSSJ(covar,mfit,nca,oneda,1,1);
for j:= 1 to mfit do
da[j]:= oneda[j,1];
if (alamda = 0.0) then
begin
COVSRT(covar,nca,mma,lista,mfit);
goto }9
end;
for j:= 1 to mfit do
begin
atry[lista[j]]:= a[lista[j]]+da[j]
end;
for J:= (MFIT+1) to MMA do
begin
ATRY[LISTA[J]]:= A[LISTA[J]]
end;
MRQCOF(x,y,sig,ndata,atry,mma,lista,mfit,covar,da,nca,chisq);
if (chisq < glochisq) then
begin
alamda:= 0.1*alamda;
glochisq:= chisq;
for j:= 1 to mfit do
begin
for k:= 1 to mfit do
begin
alpha[j,k]:= covar[j,k]
end;
glbeta[j]:= da[j];

```
```

                a[1ista[j]]:= atry[1ista[j]]
            end
        end
    else
    begin
        alamda: = 10.0*alamda;
        chisq:= glochisq
    end;
    99:
end; \{end of procedure MRQMIN\}

```

```

    procedure GETDATA;
        var
        j,k,nfiles,nobs: integer;
    begin
        i:= 0 ;
        ndata:= 0
        writeln('Enter number of input file pairs to be read :');
        readln(nfiles);
        for \(j:=1\) to nfiles do
            begin
                writeln('Enter name of brightness input data file: ');
                readln(infilename);
                writeln('Enter name of GEI vectors input file : ');
                readln(vecfilename);
                writeln('Enter \# of data points : ');
                readln(nobs);
                open(infile,infilename,old);
                reset (infile);
                open(vecfile, vecfilename, old):
                reset(vecfile);
                for \(k:=1\) to nobs do
                    begin
                    ndata: = ndata+1;
                            \(i:=i+1\);
                            readln(infile, X[i], Y[i], sig[i],alt[i],za[i],sza[i]);
                            read (vecfile, time[i], pv[1,i], pv[2,i], pv[3,i],
                                    \(\operatorname{sv}[1, i], \operatorname{sv}[2, i], \operatorname{sv}[3, i], \operatorname{vv}[1, i], \operatorname{vv}[2, i], \operatorname{vv}[3, i]) ;\)
                            zA[i]:=ANGLE (pv[1,i], pv[2,i], pv[3,i], vv[1,i],vv[2,i],vv[3,i]);
                    end; \{end of one file nobs read loop\}
                close(infile);
                close(vecfile);
            end; \{end of nfiles loop\}
        writeln;
        readln(outfilename);
    end; \{end of procedure GETDATA\}
    ```

```

    begin \{ main body of program\}
    for \(I:=1\) to 3 do
            for \(j:=1\) to 3 do
                dummy[i,j]:= 0 ; \{set dummy to zeros\}
    Xs[1]:= 0.18e-18; (total O XSections Richards \& Torr JGR 1988\}
    \(\mathrm{Xs}[2]:=1.3 \mathrm{e}-18\);
    Xs[3]:= 3.0e-18;
    Xs[4]:=4.8e-18;
    ```
```

Xs[5]:= 5.9e-18;
Xs[6]:= 6.8e-18;
Xs[7]:=6.5e-18;
Xs[8]:= 7.3e-18;
Xs[9]:= 7.3e-18;
Xs[10]:= 8.0e-18;
Xs[11]:= 9.1e-18;
Xs[12]:= 9.3e-18;
Xs[13]:= 10.0e-18;
Xs[14]:= 11.0e-18;
Xs[15]:= 11.0e-18;
Xs[16]:= 12.0e-18;
Xs[17]:= 12.0e-18;
Xs[18]:= 12.0e-18;
Xs[19]:= 12,0e-18;
Xs[20]:= 12.0e-18;
Xs[21]:= 12.0e-18;
Xs[22]:= 12.0e-18;
Xs[23]:= 10.0e-18;
WXs[1]:= 0.0373e-18;{O+(2P) R\&T '88 + Torr,Photchem.of Atmos.1985}
WXs[2]:= 0.276e-18;
WXs[3]:= 0.654e-18;
WXs[4]:= 1.08e-18;
WXS[5]:= 1.35e-18;
WXs[6]:= 1.59e-18;
WXs[7]:= 1.50e-18;
WXs[8]:= 1.74e-18;
WXs[9]:= 1.74e-18;
WXs[10]:= 2.01e-18;
WXs[11]:= 2.30e-18;
WXs[12]:= 2.35e-18;
WXs[13]:= 2.61e-18;
WXs[14]:= 2.99e-18;
WXs[15]:= 2.96e-18;
WXs[16]:= 3.18e-18;
WXs[17]:= 3.13e-18;
WXs[18]:= 3.06e-18;
WXs[19]:= 3.08e-18;
WXs[20]:= 2.99e-18;
WXs[21]:= 2.93e-18;
WXs[22]:=2.92e-18;
WXs[23]:= 0.469e-18;
XsN2[1]:= 0.60e-18;{total N2 XSections Torr Photchem. of Atmos. 1985}
XsN2[2]:= 2.32e-18;
XsN2[3]:= 5.40e-18;
XsN2[4]:= 8.15e-18;
XSN2[5]:= 9.65e-18;
XsN2[6]:= 10.60e-18;
XsN2[7]:= 10.08e-18;
XSN2[8]:= 11.58e-18;
XSN2[9]:= 11.60e-18;
XSN2[10]:= 14.60e-18;
XsN2[11]:= 18.00e-18;
XsN2[12]:= 17.51e-18;
XSN2[13]:= 21.07e-18;
XSN2[14]:= 21.80e-18;
XsN2[15]:=21.85e-18;
XSN2[16]:= 24.53e-18;
XsN2[17]:=24.69e-18;
XSN2[18]:= 23.20e-18;
XsN2[19]:= 22.38e-18;
XsN2[20]:= 23.10e-18;

```
```

XsN2[21]:= 23.20e-18;
XsN2[22]:= 23.22e-18;
XsN2[23]:= 29.75e-18;
flux[1]:= 0.117; {Shape of modified F74113 reference spectrum}
flux[2]:= 0.044;
flux[3]:= 0.700;
flux[4]:= 0.457;
flux[5]:= 0.067;
flux[6]:= 0.037;
flux[7]:= 0.257;
flux[8]:= 0.117;
flux[9]:= 1.000;
flux[10]:= 0.143;
flux[11]:= 0.094;
flux[12]:= 0.047;
flux[13]:= 0.056;
flux[14]:= 0.041;
flux[15]:= 0.043;
flux[16]:= 0.066;
flux[17]:= 0.104;
flux[18]:= 0.186;
flux[19]:= 0.051;
flux[20]:= 0.078;
flux[21]:= 0.228;
flux[22]:= 0.050;
flux[23]:= 0.033;
GETDATA;
writeln(' First variable in list to be 1)H 2)Iinf or 3)[0]@250 ');
readln(lista[1]);
writeln('Second variable " " " " 1)H 2)Iinf or 3)[0]@250 ');
readln(1ista[2]);
writeln(' Third variable " " " " 1)H 2}Iinf or 3)[0]@250 ');
readln(1ista[3]);
writeln('Enter the \# of parameters to be adjusted: ');
readln(mfit);
writeln('Enter initial H : ');
readln(A[1]);
writeln(' initial Iinf : ');
readln(A[2]);
writeln(' initial [0]@250: ');
readln(A[3]);
A[3]:= ln(A[3]);
writeln(' assumed [N2]@250: ');
readln(N2250);
writeln('N2 quenching coef kN2: ');
readln(KN2);
writeln(' O quenching coef kO : ');
readln(KO);
writeln('number of altitude intervals for BRIGHT: ');
readln(maxhindx);
writeln('altitude interval in kilometers: ');
readln(deltah);
chisqo:= 1e20;
deltachisq:= 1e20;
alamda:= -1.0; {initialization with negative alamda}
I:= 0;
while (deltachisq > 0.1) do
begin
I:= I+1;
MRQMIN(x,y,sig,ndata,a,3,lista,mfit,covar,alfa,3,chisq,alamda);
writeln(' H= ',A[1]);
writeln(' Iinf = ',A[2]);

```
```

    writeln{' [O]@250= ', exp(A[3]));
    writeln('chi sqred = ',chisq);
    writeln('end of ',I,'th iteration');
    writeln('alamda= ',alamda);
    writeln;
    if (chisq < chisqo) then
        begin
        deltachisq:= chisqo-chisq;
        chisqo:= chisq;
        end; {end of if }
    end; {end of while}
    alamda:= 0.0; {set alamda to zero for final call}
MRQMIN(x,y,sig,ndata,a,3,lista,mfit,covar,alfa,3,chisq,alamda);
writeln('alpha matrix:-');
writeln(alfa[1,1],alfa[1,2],alfa[1,3]);
writeln(alfa[2,1],alfa[2,2],alfa[2,3]);
writeln(alfa[3,1],alfa[3,2],alfa[3,3]);
writeln;
writeln('covariance matrix:-');
writeln(covar[1,1], covar[1,2],covar[1,3]);
writeln(covar[2,1],covar[2,2],covar[2,3]);
writeln(covar[3,1], covar[3,2], covar[3,3]);
writeln;
writeln(' H = ',A[1]: 7,'+/-',sqrt(covar[1,1]): 7);
writeln(' I* = ',A[2]: 7,'+/-''qrt(covar[2,2]): 7);
writeln('[O]@250= ', exp(A[3]): 7,'+/-',100*(exp(sqrt(covar[3,3]))-
1.0): 7,'名');
0500:= exp(A[3])*exp((250.0-500.0)/A[1]);
writeln(' [0]@500= ',0500: 7);
0500:= exp (A[3])* exp ((250.0-500.0)/(A[1]+sqrt(covar[1,1])));
writeln(' [0]@500+= ,,0500: 7);
0500:= \operatorname{exp}(A[3])*exp((250.0-500.0)/(A[1]-sqrt(covar[1,1])));
writeln(' [0]@500-= ',0500: 7);writeln;
writeln('chi sqred = ',chisq);writeln;
writeln;
open(outfile,outfilename, new);
rewrite(outfile);
for i:= 1 to ndata do
begin
write(outfile,X[i],'',fit[i],' ',y[i],'');
writeln(outfile,sig[i],' ',za[i],' ',SZA[i]);
end; {end of read loop}
close(outfile);
end. \{end of program TWIFITTER\}

```
```

