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LIFT THERORY OF SUPPORTING SURFACES

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## LIFT THEORY OF SUPPORTING SURFACES

Second Article
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## LIFT THEORY OF SUPPORTING SURFACES.*

Second Article
By
R. Von Lises.

This continuation of my remarks on the hydrodynamic lift theory of supporting surfaces** relates ciniefly to the task of finding suitable wing sections for given lift conditions. It is based on the formulas (set forth in wy first article for sections in general) for lift magnitude and moment, which make it possible to obtain solutions of great simplicity and clearness. For instance, every section has a point for which the lift moment is independent of the incidence. The curve inclosing all the positions of the lift resultants is usually a parabola, which can be reduced to a cluster of rays. Within these limits, there are sections, with fixed pressure center, which are consequently removed from the general class of sections by a single condition.

In the second section, I complete and correlate the valuations, proceeding from the modern theory of functions (the circle of ideas of the "distortion laws"), for the parameter determining the lift, and explain their significance in the case of the delineation of the arc. The third and fourtin sections treat of the application of the theory to the case of the Joukowski Karch 15, 1920, pp. 68-73, and Larch 31, 1920, pp. 87-89.
** First article, November 29, 1917, pp. 157-163. This article will henceforth be referred to as "I'.
section construction and its extension by Von Karman and Trefftz (this publication, 1918, pp. 111-116). The fifth section outlines a common procedure, illustrated later by examples, which makes it possible to discover any desired number of parametric groups of sections, or of suitable delineation functions. In this connection, there is a considerable mathematical difficulty, the question of the criterion for the "smoothness" of the delineation, which vas gotten around by a simpie artifice adapted to; its graphic execution, in the formulation of which I enjoyed the valuable assistance of Dr. L. Bieberbach in Frankfort. It, moreover, demonstrates in perfect harmony with experience, that the S-shaped upward bend of the section, near the following-edge, lessens the variation in the center of pressure, or may eliminate it altogether.

In a third and concluding article, I expect to take up the problem of finding the determining parameter for a given wing section.

## 1. Magnitude and Locati on of Lift.

In $m y$ first article, among other things, inciuaing tine customary postulations of eddy-free horizontal motior, the following Was demonstrated. Given any wing section containing a point or angle (Fig. I). At the angle $\alpha$ with an $x$-axis rigidy connected With the section (angle of incidence), the section is struck by an air current which has the velocity $u$, and the specific mass $\mu$ that is $\frac{\gamma}{\xi}$, or approximately $1 / 8$, under normal conditions.

There then hold, for the magnitude and moment of the lift (the resulting pressure perpendicular to the direction of the air current), simple formulas independent of the section shape, into which the section itself enters with only a few parameters. The equation for the power magnitude (Compare 29 in Art. I:) then, reads:

$$
\begin{equation*}
A=4 \pi \mu u^{2} a \sin (\alpha+\beta) \tag{1}
\end{equation*}
$$

 With reference to a given origin $\mathbb{M}$, whose coordinates thus belong also to the parameters of the section (Compare 37 in Art. I.), reads:

$$
\begin{equation*}
V_{1}=2 \pi \mu u^{2} c^{2} \sin 2(\alpha+\gamma) . \tag{3}
\end{equation*}
$$

with $c^{2}$ and $\gamma$ as additional parameters. The origin for the moment to which equation (2) refers, was called the "middle point of the section" and that direction of the attacking air (determined by $\alpha=-\beta$ or by $\alpha=-\gamma$ ) for which respectively $A$ or $M_{2}$ vanishes, was called respectively "first" or "second"axis of the section" (Art. I, section 7).

Still greater simplicity and clearness can be obtained by choosing, irstead of $u$, a new reference point $F$, mitien is found in the following manner. From we measure off a distance $U F=\frac{c^{2}}{a}$ or a line which is turned toward the $x$-axis in the positive direction by $2 \gamma-\beta$, so that the second axis of the section bisects the angle between the first axis and UF. In Fig. 1, $\beta>2 \gamma y$ was taken and accordingly $\beta-2 \gamma$ was represented as turning in the negative direction. The moment with reference to $F$, which we may now call $M$, is obtained by adding to $M_{2}$ the
moment of any power $A$ considered as affecting $\mathbb{M}$, and hence with reference to the direction of rotation.

$$
M=M_{1}-A \frac{C^{2}}{a} \cos (\alpha-\beta+2 \gamma)=
$$

$2 \pi \mu u^{2} c^{2}[\sin (2 \alpha+2 \gamma)-2 \sin (\alpha+\beta) \cos (\alpha-\beta+2 \gamma)]:$

The second expression in the square brackets can be reduced, according to the "identity" $2 \sin \varphi \cos \psi=\sin (\varphi+\psi)+\sin$ $(\varphi-\psi)$, to $\sin (2 \alpha+2 \gamma)+\sin (2 \beta-2 \gamma)$, hence

$$
\begin{equation*}
M=-2 \pi \mu u^{2} \alpha^{2} \sin 2(\beta-\gamma) \tag{3.}
\end{equation*}
$$

This equation no longer contains the variable incidence a. We have hereby demonstrated the proposition that: For every wing section there is a certain point for which the lift moment is independent of the incidence. We Wili cail this point the "focus $\because$ of the section."

From (1) and (2) is calculated the lever arm $h$ of the lift to

$$
\begin{equation*}
h=\frac{M}{A}=-\frac{\alpha^{2} \sin 2(\beta-\gamma)}{2 a \sin (\alpha+\beta)}=-\frac{h_{0}}{\sin (\alpha+\beta)}:: \tag{4}
\end{equation*}
$$

When

$$
\begin{equation*}
h_{0}=\frac{c^{2}}{2 a} \sin 2(\beta-\gamma) \tag{0}
\end{equation*}
$$

is taken. Since the lever arm of $F$ mist be extendedi in the direction of the attacking air (Fig. 2), ho represents the magnitude of its projection on the line perpendicular to the first axis. The projection of the lever arm on this line is therefore unchangeable, or: The feet of the vertical lines, from the "sec-
tion focus" to the atoucking lines of the lift, lie in a straight ine, which, at a distance $h_{0}$ from $F$, is parallel to the first axis of the section and, moreover, bisects the distance $M F$ of Fig. 1. Therewith we have reduced the determination of the attacking line for any given angle of incidence to the simplest imaginable construction.

It is z well known characteristic of ordinary porabolas, that the "foot-point-curve" for their focus coincides with the vertical tangent. Anyone tho is not familiar with this fact may make the following caiculation. For a system of coordinates with the origin at $F$ and with F'F for the x-axis (Fig. 2), the equation for the attacking lime A reads:

$$
x \sin (\alpha+\beta)+y \cos (\alpha+\beta)=h=-\frac{h_{0}}{\sin (\alpha+\beta)}
$$

and simplified:

$$
-x \cos 2(\alpha+\beta)+y \sin 2(\alpha+\beta)=-2 h_{0}-x
$$

Differentiated according to $2 x$ :

$$
x \sin 2(\alpha+\beta)+y \cos 2(\alpha+\beta)=0
$$

and with $\alpha$ eliminatedi from this and the preceding equation by squaring and adding, we obtain:

$$
\begin{equation*}
x^{2}+y^{2}=\left(2 h_{0}+x\right)^{2} \text { or } y^{2}=4 h_{0}\left(x+h_{0}\right) \tag{5}
\end{equation*}
$$

Hence: The possible positions of the attacking lines of the lift generally inclose a parabola, whose focus is $F$, whose parameter is $2 h_{0}$ and whose axis is perpendicular to the first axis of the section.

With $h_{0}=0$, the parabola is reduced to the bundle cf rays through $F$. This is the case of the vanishing variable pressure point. There is a fixed middle pressure point, through which the resultant lift always passes, the so-called "diving moment" (for $\alpha=-\beta$ ) being 0 . It is noteworthy that the section with a fixed pressure center is separated from all other sections by only a single equation $h_{2}=0$. Thereby the disappearance of $h_{0}$ (as shown by equation 4) shows that $\beta=\gamma$, or that the first and second axes of the section are parallel. In the quantity $h_{0}$ we have a definite measure of the pressure point variation (or migration). It is readily seen, for example, from equation (2), that the straight line drawn through the "center" $M$ in the direction of the second axis and, barring exceptional cases, also the straight line drawn at right angles to it, touch the parabola*:

## 2. Section Parameters.

Only five of the six section parameters, considered in my first article, are found to be essential, since $c^{\text {i }}$ and $\gamma$ appear.


* General principles concerning the location of lift resultants are also given by R. Grammel, in "Die hydrodynamischen Grumlagen des Fluges" (The hydrodynamic principles of flight), Brawnschweig, 1917, p.l3. Here the task is reduced to the determination of a "moment and center of gravity of the circulation." I cannot understand the deductions in this book. It can, however, be demonstrated that $a$ "moment of the circulation, $n$ as a quantity independent of the integration method, can only exist, when the integration is made conceming closed stream (or level) line. Now, since the lift-producing current comes out of infinity and goes into infinity, these consiaerations are of little value.
quantities may be regarded asdetermining factors for the lift relations of a section: for the moment of the lift, the length a and the angle $\beta$ (direction of the first axis); for the moment or location of the lift, the coordinates of the sectionfocus $F$ and the length $h_{0}$ or, instead of the last, the location of the first axis (which passes through $M$ ) at the distance $2 h_{0}$ from $F$ (base line of parajola). How these five quantities are to be ascertairuci iuz $\dot{\sim}$ my first articie, and $I$ will indicate it briefly here.

If we regard the coordinates $x$ and $y$ of a point in the plane of the section as components of a complex quantity

$$
z=z+i y
$$

then there is one, and only one, evolvable complex function $\zeta$ of $z$ in the form (Equation 7 in Art. I).

$$
\begin{equation*}
\zeta=z+\frac{a^{2}}{z}+\frac{a_{z}}{z^{2}}+ \tag{6}
\end{equation*}
$$

through whici any point of a circle is coordinated with every point of the section perimeter and any point outside tine circle (7), with every point. outside the section, in constant and definitely convertible manner. The radius $a$ of the circle and the coordinates (contained in $m$ ) of its midile point $M$ are through

$$
\begin{equation*}
\zeta-m=a \tag{7}
\end{equation*}
$$

the coefficients of $a$, $a$..... cf (6), definitely determined, aslikewise every point $B$ of the circle into which the section tip is drawn. Thereby the parameter values are found. The radius $\underline{a}$ is the parameter referred to above under the same desig-
nation. $B M$ is the first axis of the section and, hence $P$ is the angle formed by BM with its x-axis. The focus $F$ is finally obtained, by making the vector $L \sim F$ equal the complex quantity $-\frac{a_{2}}{a} e-\beta i$ (according to Equation 34 in Art. I and the above procedure).

In order to obtain a cursory survey of the parameter values and of the location of the invariable points and straight lines, We will next consider the arc as the approximation, or better, as the limit, of a wing section. Strictly considered, the "infinitely thin" arc does not fall under our conception of a section, because it does not possess simply one place of variable tangent direction, but two of them. We car only dispose of it in the sense of our consideration, by disregarding ore edge (the right one, for example) and not leiting ourselves be disturoed by the fact that the calculated current is unrealizable on this edge. The resuits are accordingly significant only in that they give the approximate position (of the focus, for example) for a very thin section, closely fitting the arc, which section has its tip in the left corner.

The complex process, which tiansforms a circle into a doubly traversed arc, is known and may be expressed in the form

$$
\begin{equation*}
z=\zeta+\frac{c^{2}}{\zeta} \tag{8}
\end{equation*}
$$

When applied to a circle passing inrough the points $\zeta= \pm f c$ ( $B, B_{2}$ in Fig. 3). Fig. (8) comefunder the form (6) desired by $u$ : since its solution according to $\zeta$, by development according to the descending powers of $z$ begins with the memors.

$$
\begin{aligned}
&-10- \\
& \xi= z-\frac{\delta^{2}}{z} \cdot \cdot . \cdot . . . . . . . .(8!)
\end{aligned}
$$

In the transformation, the end points $z= \pm 2 c$ of the arc correspond to the points $\zeta= \pm c$ on the $x$-axis. To any given point A of the circle (Fig. 3) and to the $A_{1}$ on the symmetrical curve (Fahrstrahl) (so that $\Varangle O A, x=\Varangle x, O A_{1}$ ), there corresponds one and the same point $A^{\prime}$, which is obtained by the addition of the vectors $O A$ ard $A_{2}$. Ficr, ascuraing to the principle of the constant "power of acircle," the length of $O A_{1}$ is like that of $O A_{2}$, like $c^{2}$ through the length of $O A$, hence with $O A=\zeta$, as the comilex equation: $O A_{1}=c^{2}$ : ५. That, in fact, the point $A^{\prime}$, found by this process, falls on an arc, is readiily seen from the foliowing:*

If the length $O A$ is represented by $\rho$ and the angle between $O A$ and the $x$-aris by $\psi$, the coordinates of $A^{\prime}$ are according to the equations:

$$
x=\left(\rho+\frac{c^{2}}{\rho}\right) \cos \dot{\psi}, y=\left(\rho-\frac{c^{2}}{\rho}\right) \sin \psi,
$$

from which we obtain:

$$
x^{2} \sin ^{2} \psi-y^{2} \cos ^{2} \psi=4 c^{2} \sin ^{2} \psi \cos ^{2} \psi
$$

On the other hand, the perpendicular $M A_{0}$ from the middie point 3 to $O A$ shows that $O A_{0}$, half the difference between $O A$ and $O A_{2}$, consequently of $\rho$ and $\frac{c^{2}}{\rho}$, has the value $s$ sin $\psi$, Then $s=O M$ denotes the distance of the center of the circle from 0 . There follows, therefore, from the expression given for $y$ :

$$
y=2 \mathrm{~s} \sin ^{2} \psi
$$

$\frac{\text { and if this value for } \sin ^{2} \text { in is substituted in the preceding }}{\text { Aner }}$ cownlex cilculation, follows from the deduction given in Sec. 4.
equation and simplified:

$$
\begin{equation*}
x+\left(y+\frac{c^{2}-B^{5}}{s}\right)^{2}=\left(\frac{c^{2}+s^{2}}{6}\right)^{2} \tag{9}
\end{equation*}
$$

as circle equation for position of $A^{\prime}$. Thereby, since $s>0$ and $\sin ^{2} \psi>0, y$ can only have positive values. The arc has the camber $2 s$, the chord $4 c$, the radius $\frac{c^{2}}{s}+s$ which is readily constructed, and the half center angle sin $\frac{2 c s}{c^{2}+s^{2}}$.

The task of finding the parameter (as also the inverse) for a given are $B^{\prime} \mathrm{B}^{\prime \prime}$, with chord 40 and camber 2 s , as the limit of a section with its tip at $B^{\prime}$, is executeci as follows. Divide the chord $B^{\prime} B_{1}^{\prime}$ into four parts at $B, O, B_{2}$ and bisect the vertical line $O S$ at $M$. Then $B$ represents the tip and $M$ the center of the section. Therewith the radius $a=\sqrt{s^{2}}+\overline{c^{2}}$ and the direction $B M$ of the first axis with $t g \beta=s: c$ are found and the lift magnitude is thus determined. In order to obtain the moment-course, after the location of the first axis is already known, we only kave to find $F$ and, for this purpose, according to ( $8^{\prime}$ ), the distance $M F=c^{2}: a$, under the angle $-\beta$, is measureA 0 ff 5 the line $M B_{1}$. $F$ is found constructively in the simplest way (Fig. 3) by erecting at $B_{1}$ the vertical Ine $s$ perpendicular to the chord and projecting its terminal point on $\mathrm{MB}_{2}$. The projected line contains, in addition to the focus F, also the center $M^{\prime}$. of the arc. The straight line $B M$ is the base line of the lift-parabola and of the parameter $h_{0}$, equal to half the distance between $F$ and $B M$.

In Fig. 4 there are represented several arcs, which are de-

Veloped fron the same circle by choosing different starting points 0, witr their axes, parabolas, etc. If the curve becomes 0 , then the "straight line" section coincides with its first axis. The focus is situated or tinis line at $3 / 4$ of the distance from the left end and the parabola is reduced to the bunch of raya, which means that all attacking lines pass through $\vec{F}$. The greater tne curvature, just so much farther $F$ moves from the section surface toward the interior, but always remains at about $3 / 4$ of the chord. In Fig. 5 the so-cailed "pressure point migration" is represented, with reference to the choris for the curvature relations $0,0.1,0.2$, and 0.3 , in the customary foru. We can see how it increases with ircreasing curvature. Since there is an endiess number of different parabolas, all possible cases within the series of arcs are in a certain sense exhausted.

For giren section forms, the paraneter vaiues, firsi knom from the theory of "consistent iritation," are suiojected to the
 ence points for their determination. If we supplement the data in my first ariicle by a restriction for the location of from the same source (Bieberbach's "Law of Surfaces") and, furiher, by the adoption of an old lim (from Lunciau and Tecplitz) as well as of a continuation accordirg to my first article, of the hyorodynamic theory by Frani and Lowner and, lastly, of a new work by G. Pick or "Consistent Imitation," we ther obiain the following determination methods.

1. For the value of the radius $a$ : The radius of the curve of a section determined by (6) is at leas equal to $1 / 4^{*}$ and at most to $1 / 2^{* *}$ of the longest diameter of tie section. For the lift coefficient, with reference to the greatest width $b$, there are obtained therefrom theinequalities

$$
\pi \sin (\alpha+\beta) \leqq \frac{A}{\mu u^{2}-} \leqq 2 \pi \sin (\alpha+\beta)(10)
$$

In the case of very narrow sections, the lower limit is only slightly exceeded.
2. For the location of the center $\mathbb{M}$ : The center of the curve of a section ("section center") is located so that the Whole sections falls within the circle described about it with the radius $2 a^{* * *}$. Furthermore, M Iies within the smallest convex space**** inclosinE the seciion (Iiko the center of gravity of a surface on the section perimeter). The first condition in the case of very narren sections is exastly fulfilled, that is, the ends of the section iie near the perimeter.
3. For the location of the focus F: The above defined. focus of the section (foous of parabola inclosing the lift attacking lines) lies within the perimeter (Bieberbach, a.e. O., prin* This statement is containsd in the first of the principles mentioned under 2 .
** Landau and Toepiitz, Arch. f. Math. and Prys. II (190E), pp. 302-307; compare aldo Frank and Lowner, "A unten a.0." *** L. Bieberbach, Sitzungsber. d. Berliner Akad. XXXVIII (1916) pp. 940-955, principie $v$.
**** Franit and Lowner, Math. Zeitschi. 3 (1919), pp. 78-86. Here the following is demonstraied: If the surfece is considered. as uniformly sovered with substance and if, in the transformation, $\epsilon$ ach point is allowei to retair its mass, the center of gravity then jemaine at M.
cipie I). In the case of narrow sections, it is not far from the eage of the circle. Its distance from the circumference is greater than $r^{2}$ : $a$, when $r$ denotes the radius of the circle inscribed in the section (G. Pick, Sitzungsber, d. Wiener Akadamie, math. raturv. Klasse, Abt. IIa, 126 (1917), pp. 247-263, equation. 4). According to the resilts obtained by L. Bieberbach and G. Pick, the first principles statea ander 1 and 2 can be made still broader: not only the section itself, but every circle which is created by concentric dupiication of a circle lying entirely within the section, falls in the duplication of the perimeter (Pick, a.a.0. eq.9). Further, when $r$ denotes the radius of a circle entirely within the section and $d$ the distance of its center from any given point of the section perimeter, then

$$
\begin{equation*}
a>\frac{1}{4} \frac{(d+r)^{2}}{d} \tag{11}
\end{equation*}
$$

while the first siatement uncer 1 only claims substantial ly that a $>1 / 4(d+r)$ (Pici., a.a.C. eq. V. and Bieberbach, Math. Amal 77 (1916), pp. 153-172). Therefrom, for example, the conclusion may be draw that a thickening of the front end (leating eage) of a section raises the lower limit of the lift coefficient. In Fig. 6, where $b$ denotes the greatest midth of the section, it is possible to construct, in an easily understocd manner, for the radius of the rounded front edge, the length $b^{\prime}=(d+r)^{2}: d$, which is greuter than $b$ and holas good for a $1 / 4 \mathrm{~b}$.

The parameter $\nabla$ alue and tierewith the iift magnitude and location are natural ly only incompletely detemined by all these
principles. The exact determination, for one of its form according to the previously given section, requires, each time, the solution of the problern of "consistent imitation," that is, the determination of the function (6) belonging to the section. My next article will take up this task. In the present article, I will only take up the inverse task of determining sections for given parameter values. We will first consider the known examples theoretical $l y$, that is, sections defined by their delineation functions.

## 3. The Joukowski Section Form.

The Joukowski section form is known to be produced, when the transformation (8) is applied to a circle $K$ which passes through the point $\zeta=-c$ ( $B$ in Fig. 7) and embraces the point $\zeta=c$, near its edge ( $B_{2}$ in Fig. 7). Since a circle (with center $M_{0}$, dashed in Fig. 7), passing through both points and hence transformable by ( 8 ) into an arc, can be so placed that it touches the given circle at $\xi=-c$, then the image of the latter, in the vicinity of $\zeta=-c$ mist appear as the left end of the arc. The joukumbixi section has, on the left, a point, which stands at an angle of arc $\sin \frac{2 c s}{c^{2}+s^{2}}$ to the direction of the chord, in which $s$ denotes the segment $O M_{0}$ of the straight line $B M$ on the axis of ordinates. Moreover, the section approaches the arc of a circle more closely, the nearer the point $B_{1}$ is to the circumference.

For the construction of the section, we note that when the point $A$, as the end point of the vector $O A=\zeta$, runs through
a circle $K$, also the end point $A_{1}$ of the vector $O A_{1}=\alpha: \xi$ must move on a circle $K_{2}$. The circle $K_{1}$ proceeds from $K$, throdgh the transformation of "reciprocal radii" and the corresponding reflection on the $x$-axis, and touches: $K$ at the point 3. Consequently, the center $\mathbb{K}_{1}$ of $K_{2}$ is found imnediately, by taiking the direction $\mathrm{OM}_{1}$. on the left symmetrically to 0 N ard locating $H_{2}$ on Bir. The poirt $A_{1}$ (enreesponding to A) of the section perimeter is found, as in the case of the arc formation, as the sum of the related vectors $O A$ and $O A_{1}$, winich afain lie symretrically with the x-axis, but now have their end points on different circles. In order to construct the complete section, after finding $K_{2}$; it is only necessary to draw through 0 a bunch of symmetrical lines, for example, all lines at intervals of $30^{\circ}$, and thus obtain as miany points of the section as desired, by simply orawing parallel lines or intersections* (Fig.7). Only the points falling on the axes must be taken $\mathrm{b}_{\mathrm{y}}$ the compasses. It is ofter useful to know the tangent direction of a system of points for constructing the curve. I wiil here give the following very simple construction without demonstration. Erect at $A$ and $A_{1}$ the periendiculars to the lines $O A$ and $O A_{1}$ and mark on them the points $N$ and $N_{1}$ so that $O N$ is parallel to Li A and Coll is paizallel to $M_{1} A_{1}$. Then $\mathrm{N}_{1} \mathrm{~N}_{1}$ gives the direction of the normal perimeter at the point $\mathrm{A}^{\prime}$.**
E. Trefftz gave a similar method of constraction. See this publication, Vol. IV, (1913), p. 131.
** The proof follows from the general theory of "geometrical differentiation," which shows how to find the tangents to any curve defined by point construction and which I developed in Zeitschr. f. Nath. in phys. 52 (1905), pp. 44-85.

For a Joukorski seotion obtained in this way from its "image circle" (Bildzreis) $y$, the lift paremeter is immediately deducicie. The radius of $K$ is the lengit in formial, and $B M$ is its direction and location with reference to the first axis, hence. the angle of this ilia ritin the x-axis of the angle $\beta$. The second axis has the $x$-direction, since, oi. ascount of the real value of $c^{2}=-a_{1}$, the angle $\gamma$ varishes. The forus $F$ is found by taking $M F=\alpha^{2}: \&$ on the liaie ecine from $E$ to tie seccrd intersection point of $X$ with tie x-axis. The best way is to take or. tinis line (as also on the line $N D$ drawn through ai parallel to the $x$-aتis) the distance $180=\mathrm{NO}^{\prime}=0$ and then draw of parallel to DC'. The oase line of the parabola is $B: L$ and therefore the parameter $h_{0}$ equals half the distance between $F$ and BM.

If it is desired to foliow dip connection betaen the pressure point migration anci the shape of the section; a goca way is to iruagine all those sections wich are ierived Erom different circles through tinis trensformation process (with fized c). Thereby the "first axis" $3: 1$ remains urohanecd, as likerise the directicn $M F$, while the distance betmeen $T$ and the axis varies according to the ratic $c^{2}:$ a, ience diminishing with the increase in the jepgth of the radius. Since the section becomes thicker with ircreasing a (the circumference receding farther and farther from the point $B_{1}$ ), it is evidunt that a thickenirs of the section lessens the pressure point migration and vice verac We have seen above that, in the arc, the distance of the focis
from the axis grows with increasire curvature. he may now say, therefore, that the survature ans thickening of the section exert a compersating effect with reference to the pressure point migration. In Fig. 8 tinere is dramn a series of three Joukomski sections which (even in their location) belong to the same lift parabola. It is seen that, while the thicimess varies greatiy, tiae ourvature (underneath) only varies slightly.

Jolikowski sections with fixed center of pressure are obtained only when the curvature (uncerneath) is zero. These are tiae club-shaped sross-sections which come into consideration for stays, etc., but not for supporting-sections.

## 4. Enlarged joukowski Sections.

Ti. v. Marman and E. Trefftz (this publication, Vol. IX, 1918, pp. 111-116) have, in connection rith a remark of Kutta (Sitzungsber d. bajer. Akad. d. Wiss. Math.-Physik. Klasse 2911; p. 77), somewhat extendea the transformation (8) applied by Joukowsiri to the construction of the section and confirmed, for the $=$ cose, by finont calculation, the gereral formilas repeated in (1) and (2) for lift magnitude and moment. Their expression for the moment is considerably more complicated, simply on account of an uniavorable choice of their reference point.

Equation ( $\varepsilon$ ), by once adding $2 c$ and once subtracting $2 c$ and finding the quotient, may be brought to the form

$$
\begin{equation*}
\frac{z-2 c}{z+2 c}=\left(\frac{b-c}{b+c}\right. \tag{12}
\end{equation*}
$$

in wioh it is seen, even without the calculation given in section 2 , that there corresponds to the circle going through $B$ and $B_{1}$ in the $\zeta$ - plang (Fig. 9) a doubly intersected arc through the points $. z= \pm 2 c$ ( $E^{\prime}$ and $B_{1}^{\prime}$ in Fig. 9). Then, Finen $A$ is anypoint (designated by $\zeta$ ) of the first-named circle, the complex numbers $\zeta \pm c$ are represented by the vectors EA respectively. $B_{2} A$, and when $A$ is coorcinated with $A$ in conformity with (3) and (12), the numbers $z \pm 2 c$ are repregented $\mathrm{by}_{\mathrm{y}} \mathrm{B}^{\prime} A^{\prime}$ respectively. $\mathrm{B}_{2}^{\prime} \mathrm{A}^{\prime}$ ! Since tive quotient of two complex numpers in the angle caiculation has the ancle of both vectors representing the rumbers, then (12) signifies that $\rightarrow B^{\prime} A_{1}^{\prime} B_{1}^{\prime}=2 \neq B A B_{1}$. Now an arc is known to be characterized by the properiy of possessing a constant angle at the ciroumference, so that tiee arc EAB' is transformed into an $\operatorname{arc} B^{\prime} A^{\prime} B_{1}^{\prime}$ of double the angle at the circumference. The loner arc $\mathrm{BE}_{2}$ of tine original circle has the angle at the circunference which supplements the upper angle to $130^{\circ}$, so that the double angles aviusuingly compine to give $360^{\circ}$ and lead. to the same arc over $B^{\prime} B_{1}$.

Equation (12) cin be chanced so that the reciprocal reIation of certain arcs remaing the same, without, however, the same arc's corresponding twice to the circle $B B_{1}$. If we write, namely,

$$
\begin{equation*}
\frac{z-c^{\prime}}{z+c^{\prime}}=\left(\frac{\zeta-c}{\zeta+c}\right)^{n} \tag{13}
\end{equation*}
$$

in which $c^{\prime}$ denotes a still-to-be-determined constant, and $n$ one of $t w o$ different exponents, then to each arc through $B B_{2}$ there is a corresponding are through $z= \pm^{\prime}$ with the $n-f o l d$. angle at the circumference. The arcs, which correspond to both parts of a circle through $B B_{2}$, consequently have angles at the circumference which differ by $n \pi$ and therefore intersect, if $n$ is a little larger or smaller than 2 , at the acute angle $\pm(n-2) \pi$. The constant $c^{\prime}$ must be determined from the condition that (13) assumes the form established in equation (6), namely, that the infinity in $z$ and $\zeta$ are identical. If the numerator and denominator are shortened in (13) by $z$ resp. $\zeta$ and developed according to $I / z r e s p . I / \zeta$, the first members read

$$
I-\frac{2 c^{1}}{z} \ldots \ldots=I-\frac{2 n c}{\zeta}
$$

so that $c^{\prime}=n c$ wust be written in order that $z=\zeta$ to infinity. In fact, with tiis value of $c$ (and with $\zeta$ developed according to the falling powers of $z$, the solution of the problem gives

$$
\begin{equation*}
\zeta=z \frac{m-I}{3} \frac{e^{2}}{z} \tag{14}
\end{equation*}
$$

Thus the coefficient $a_{1}$ of $I / z$ is real, so that $\gamma=0$, that is, the "second axis" of the section has the direction of the x -axis.

The crescent, into winich the circle through $\mathrm{BB}_{2}$ is converted by the transformation (13), represents no section in our sense, because it presents not one but two "singular" locations. But we can, in a very $s$ imilar manner to that of the Joukowski
trarsformation, cotair a correct section, by applying (13) to a circle which contains only $B$. on its circumference and $B_{2}$ inside near the circumference. These are the sections which Von Karman and Trefftz considered. In Fig. 10, a section of this kind is drawn, with $n=1.95$ and with the aid of a calculated point. The shape of the perimeter is, in a general way, similar to the Joukowski section, excepting that the left end doesnot represent a point but a very acute angle, even with the angle $-(n-2) \pi=$ $0.05 \pi=9^{0}$.

The construction of the section, with the small values of $n-2$, is not very simple. It is best accomplished with the aid of two groups of circles, as suggested by Karman and Trefftz. Of the parameters, which determine the lift, the length a, as the radius of the base circie; then the "first axis," according to the location and direction of. BM ; and, lastiy, the direction $B O$ of the second axis (real $a_{1}$ ) are given directiy. In order to find the focus $F$, we must take the length

$$
M n=-\frac{a}{a}=\frac{n^{2}-1}{5} \frac{c^{2}}{a}
$$

from the center $M$ at the angle - $\beta$ tims on the siraight line, which goes from $M$ to the second intersection point of the circle with the x-axis. Thereby the lift relations are fully determined. Reversed, re, can see how (with the asammption of a straight line $B$ in as base line and a point $F$ as focus of the lift parabola for the given $a, \beta$ and $n$ ) to find the lecation of $M$ and $B$. and from them to construct the section. In Fig. 10, the lift parabola is shom for the traced section.

If we wish to estimate the dependence of the pressure point migration on the incidence of the section, we must vary the value of $N$, without varying the oircle. There then remain stationary both the "first axis" BM and the line $\mathbb{N B}_{1}$, on which $F$ is located, and the parameter $h_{0}$ of the lift parabola increases in proportion to $3 \mathbb{N}$ and consequently to $n^{2}-1$. But now it must be remembered that the lengil of the section increases approximately with $n$, because witi $c \cdot=$ nc, the abscissas of the single points of the figure vary in this proportion. Hence, the comparative change in the pressure point migration is determined by $\frac{n^{2}-1}{n}$ and it will be advantageous to let $n$ fail below the Joukowski vaiue. With an incidence of $12 \%$, we have $n=2-1 / 15$ and therefrom the change of the pressure point migration $\frac{n^{2}-1}{n}$, correspording to an improvement of $5.4 \%$.

In connection with the above, we may now conclude that lessening the curvature, increasing the thickness and increasing the incidence exert a favorable influence on the pressure point motion. Sections with fixed center of pressure are obtained nere, however, as in the original Joukowski case, oniy with the curvature zero, namely, with symmetrical "stay" sections.
5. General Procedure.

The Joukowski section construction, including the above mentioned extension, can only be oonsidered as a very special exarnple of sections, for which the general lars of lift hold good, as set forth in section 1. Three characteristics (curvature, thick-
ness and incidence) can indeed be varied at will, but in contrast with the possible multiplicity of shapes, only a very limited number has been obtained. As sections with fixed pressure center, for example, there have been obtained only the symaetrical shapes unsuited for supporting surfaces. Finally, its whole development, however valuable and fruitful it may have been for building up the theory, shows, in its reliaition tc tie seeoinl are portrayal a one-sidedness which is not grounded in the nature of the problem and from which we must be completely freed, if we are to obtain the full value of the theory.

We start out with the supposition that the reverse of each delineation of the form (6) applied to a circle (7), if the coefficients $a_{1}, a_{2}, a_{3}, \ldots$. only fulfill certain conditions, must lead to the section of a supporting surface. These conditions are: 1. The circle (7) must be converted into a simple, closed, double-point free curve, which answers the common form requirements which may be made of a supporting-surface section; 2. The portrayal must in the whole oursiac space of the circle, be simple, that is, reversible; 3. To one point of the circle there must correspond, in any given case, a point or angle of previously determined opening; 4. The lift-determining parameters must have approximately predetermined values.

The last point is most simply executed, after the preceding one. For, when the circle, whose image the section is to be, and, on it, the point $B$, which is to be converted into the section tip, is arbitrarily chosen, there only remains the location of the
focus, for deternining the lift. Tis location dependis, however, accoraing to section 2, only on the first coefficient $a_{1}$, which is thus alone partially determined. With regard to condition 3, it need only be borne in mind that at the chosen point $B$, the deduction $\frac{\dot{\alpha} z}{\alpha \zeta}$ must vanish, which signifies an easily prescribed condition for tie coefifictents, especially when this disappearance is of a definite order. The real deciding difficulty lies in condition 2 , since it forms a known and rot yet fully solved problem of the theory of furctions, for recognizing the reversibility of the delineation on the properties of the coefficients. We can, nevertheless, easily get around this difficulty; and indeed in connectior with the consideration of condition 1. Naturally, one can not make sure of the shape of a section, without actually drawing it once. From the nature of the thirg, it is not possiole to give definite rules; without first being obliged to try out one or another 上ypothesis. Therefore it would be of no great practical importance for us to be able to know positively, in advance, that the corstruction will give a double-point ine. Our solution now consists therein tiat me, in the decision on the reversibility of the diagram, proceed on the assumption that the section perimeter, and hence the representation of $K$, has already been drawn and found to be free from the double-point. The followiñ staterient (whose simplification, as compared with my original proposition, is due to a friendly suggestion of Prof. Bieberbach) therefore holds good. As the reverse of (6) let

$$
z=\zeta+\frac{c_{2}}{\zeta}+\frac{c^{2}}{\zeta^{2}}+\ldots
$$

a representation, which converts the circle $\zeta-m=a$ into a simple closed double-point free line. The zero points of $\frac{d z}{d \zeta}$ may all lie within the circle, excepting the one that lies on the circumference itself. Then the representation in the whole outside space of the circle is reversibie, when the origin $\zeta=0$ falls inside the circle.* As to how the principle is applied will be immediately evident in the description of the whole construction process.

We will confine ourselves to the case in which the incidence is zero at the rear end of the section, which accordingly constitutes a true point. In this case, the deviations of the entire section, in comparison with one having an angle of 6 to $12^{\circ}$, are practically negligible. Loreover, our method can be employed, even for the more comrion asamption, as soon as, instead of $\zeta$ and 2 , the corresponding powers of these variables or their combinations are properly applied.

The coordinates are so placed in the $\zeta$ plane that the circlepoint $B$, which is converted into the section tip, has the negative abscissa - $c$ and the ordinate zero, astin the Joukowski ex* The principle may be made intelligible as follows: From the given condition, in harmony with the fact that in infinity $z$ is opposed. to $\xi$, it follows that a sufficiently large circle K', con centric with $K$, is formed with a double-point-free circumference $P^{\prime}$ of any desired size, in which some point outside of $K$ corres ponds to every point outsiae of $P$. Now, any desired point $z$, which lies between the section outline $P$ and the outer circumfer ence $P^{\prime}$ can always be converted into a point in the space outside of $P^{\prime}$. Thereby the number of the points, coordinated with $z$ out. side of $K$, cannot be changed, because there are no branching points of (15) in the space outside of $K$ and because no passage of the image point through $K$ can take place, so long as $z$ does not pass through $P$. An exact proof for the given case com be fig-
ample. The circle $K$ which is to produce the section, must therefore, pass through this point $\zeta=-c$, while inclosing the point $\zeta=0$, The delineation function $f(\zeta)$, which above was a binomial (8), we now write in $n+1$ terms.

$$
\begin{equation*}
z=\xi+\frac{c_{2}}{\zeta}+\frac{c_{2}}{\xi^{2}}+\ldots \frac{c_{7}}{\xi^{2}} \tag{16}
\end{equation*}
$$

The few limitations, to he manie fas the $c_{1} \ldots c_{n}$, gill leave sufficient play for their application to the most widely differ ing reguirements. We must next find the deduction from (16)

$$
\begin{equation*}
\frac{d z}{d \zeta}=I-\frac{c}{\zeta^{2}}-\frac{20}{\zeta^{3}}-\cdots-\frac{n c_{n}}{\zeta^{n+}} \tag{17}
\end{equation*}
$$

The right side is a product of $n+1$ factors of the form ( $1-\frac{V}{\zeta}$ ), in which one of the $\nabla^{\prime} s$ must have the value $-c$, while the total of the $\nabla^{\prime} s$ is subject to the limitation, that their sum vanishes, so that, in the product, the member with $\frac{I}{\zeta}$ drops out. Accordingiy, we proceed so that we (instead of imruediately choosing the quantities $c_{1} \ldots c_{n}$ ) adopt the points $\nabla_{2}$, v... $v_{n}$ Witinin the circle $K$ with the contraetion.

$$
\begin{equation*}
\nabla_{1}+v_{2}+\ldots v_{n}=c \tag{18}
\end{equation*}
$$

(Fig. 11 with $n=2$ ) and from these chosen $v^{\prime} s$ construct the polynomial (17), as follows:

$$
\begin{array}{r}
\left(I+\frac{c}{\zeta}\right)\left(I-\frac{\nabla_{1}}{\zeta}\right)\left(I-\frac{\nabla_{2}}{\zeta}\right) \ldots \ldots\left(I-\frac{\nabla_{n}}{\zeta}\right)- \\
I-\frac{c_{1}}{\zeta}-\ldots \frac{n c_{n}}{\zeta+1} \cdot \cdots \tag{19}
\end{array}
$$

(Continued from p.24)
ured out by analogy with Bieberbach's demonstration (Funktionenthe orie, Sammg. Goschen, No.768, p.56) of a similar p=oposition.

The values of the individual quantities $c_{1} \ldots c_{n}$ are obtained by putting like powers on both sides, for example:

$$
\begin{align*}
-c_{2}=v_{1} v_{2} & +\ldots v_{n-1} v_{n}-c\left(v_{2}+v_{2}+\ldots v_{n}\right)= \\
& =\Sigma v_{l} v_{k}-c^{2} \ldots \ldots \tag{20}
\end{align*}
$$

Hereby the following is accomplished. The point $R$ is converted into a tip and all other junction points lie within the circle. Then the delineation function has the form (16), that is, the coefficient of the linear member is 1 and that of the comstant is zero. If the amount of the pressure point migration, and hence of the parameter $h_{0}$ of the lift parabola, is predetermined, then the quantities $\nabla_{1} \ldots v_{n}$ must only be subjected to the second restriction, in order that the complex quantity $\frac{c_{2}}{a} c^{-\beta i}$ (in which a represents the radius, $\beta$ the "angle-argument" of the distance $-c+m$ from $B$ to the center $M$ ), drawn from the center $M$ to $F$, may have the desired distance $2 h_{0}$ from $B M$.

Naturally, it is not necessary to take the circle $K$ and distribute the points $\nabla_{2} \ldots v_{n}$ within it in advance. It is better, namely, when $h_{0}$ is predetermined, to make first an assumption for the first values of $n-1$ of the $\nabla$, then to calculate the nth value from (18), and then $c_{2}$ from (20) and lastly choose the circle so that it will inclose the $V$ 's and will satisfy the conditions with respect to the foms $F$. If, for instance, we wish to obtain a section with a fixed pressure center, then it is only necessary to locate the center $M$ of the circle, so that BM will have the direction of $\sqrt{c_{2}}$ (that is, $\beta=\gamma$,

With the first and second axes parallel).
After the circle $K$ and all the $V^{\prime}$ s have been determined, then the section outline can be located point for point, as the representation of K . Thereby it is to be recommended, namely, for the first trial, to represent only a few points of the circle (measured from 0 ) about $30^{\circ}$ apart. The individual summands" of (15) are comparatively easy to determine and to combine, partly by calculation and partly by construction, as we will further illustrate by examples.

The chief practical difficulty always consists in finding such values for the $v$ 's that utilizable shapes will rear lt from them. To this end, a certain skill can be acquired through the execution of many examples. A reference point may be suggested by the remark that the crowding of the V's in the vicinity of the zero point (excepting the one which on account of "18". must lie at $\zeta=c$ ) leads to the Joukowski forms.

## 6. Examples. Sections With Pressure Center.

As the first example, we will construct a section of the form $n=3$, Without any previous instructions in regard to the lift values. We accordingly choose (Fig. i2), on the real axis, two points $B$ and 0 , at a distance $c=0.4$ (a length of 100 mm . was taken as the unit in the original figure) and a circle $\mathbb{B}$ going through $B$ and inclosing $O$ with the center $\mathbb{L}$ and radiius $a=0.44$. The angle included between $B X$ and the axds is the angle $\beta$ of Section 1 . BM is the first axis of the section
and $M$ its "center". We choose the three $\nabla-\nabla$ alues as follows:

$$
v_{1}=c, v_{2}=-v_{3}=\frac{c}{2} e^{\frac{\pi i}{4}}
$$

so that they satisfy equation (18) and, as shown by Fig. 22, the end points $V_{1}, V_{2}$ and $V_{3}$ of the vectors $O V_{1}=V_{1}$ etc. lie within K. Equation (19) is now converted into

$$
\left(I-\frac{c^{2}}{\zeta^{2}}\right)\left(I-\frac{\nabla_{2}^{2}}{\zeta^{2}}\right)=I-\frac{c_{1}}{\zeta^{2}}-\frac{2 c_{2}}{\zeta^{3}}-\frac{30_{3}}{\zeta^{4}}
$$

from which proceeds

$$
c_{1}=c^{2}+v_{2}^{2}=c^{2}\left(I+\frac{1}{4}\right), c_{2}=0, c_{3}=-\frac{c^{2} v_{2}^{2}}{3} \frac{c^{4} 1}{I 2}
$$

Hence the transformation equation reads:

$$
2=\zeta+\frac{c_{1}}{\zeta}+\frac{c^{3}}{\zeta^{3}}=\zeta+\frac{c^{2} I+\frac{i}{4}}{\zeta}-\frac{c^{4} i}{I 2 \zeta^{3}}
$$

in which $c=0.4$ is to be introduced. The solution of this equation is best accomplished, partly by calculation and partly by construction.

We next determine the value of $c_{1}$ and the direction of $c_{2}$. For this purpose we took, in Fis. $12, \quad B C=-\frac{c}{4} i$, $\overline{\text { G }}$ that $O C=-\frac{C_{1}}{c}$, and bisected the angle $B O C$, whereby the angle $\gamma$ of Section 1 and therewith the direction of the second axis of the section was found. Counting from this second axis, the racial lines $\zeta$ were drawn at every $30^{\circ}$ toward the points $A$ of the circle $\mathbb{K}$ (only one is actual ly traced, simply the terminal points of the others being markedi). The end points of the radial Iines $c_{2}$ : $\zeta$ lie, according to the known characteristic of transformation through "reciprocal radif," again on a circle $\mathbb{K}_{2}$. Its
center $M_{2}$ lies on a straight line $\mathrm{OM}_{2}$, Which is situated symmetrically to OMS with reference to the perpendicular to the second axis. One point of $X_{1}$ is $C$ and another one can eas if be obtained from the proportion

$$
\zeta: c=0 c: \frac{c_{1}}{\zeta}
$$

After $X_{2}$ has been obtained, the points $\xi+\frac{c_{2}}{\xi}$ can be located by combining the symmetrically situated vectors $O A$ and $O A_{1}$, just aspn the Joukowski case. In Fig. 12, the construction is completely carried out for one point and the rest of the 12 points of $K_{2}$ are simply indicated, so far as the size relations allow. Now each point $A^{\prime}$ must be shifted $c_{3}: \zeta^{3}$. The direction of $c_{3}$ is that of the negative imaginary axis. Hence the shifted: direction is obtained approximately for the point with the amplitude $\gamma-30^{\circ}$, by taking the angle $180-3 \gamma$ from the real axis. The construction is easily completed for the remaining points: since the direction changes in each instance by $3 \times 30=90^{\circ}$. On account of its smellness, the length of the displacement is best determined by calculation. It is $c_{3}=0.00213$ and hence for the just named point $A$, with $\zeta=0.514$, the length $A^{\prime} A^{\prime \prime}=0.0156$ and we must take $A^{\prime} A^{\prime \prime}=2.56 \mathrm{~mm}$. according to the chosen unit of measure. Thus 12 circumference points are definitely determined in Fig. 12, as also, in very simple fashion, both the points, right and left which proceedi from the intersection points of $K$ with the real axis.

If we draw through $M$ the parallel to the second axis and transfer its angle with the first axis to the lower side of this
parallel, we thus obtain the direction $M F$ toward the focus of the section, on which $M=c_{2}: a=0.375$ is to be taken. With $F$ asithe focus and $B N$ as the base line, the lift parabola is determined. It is seen that the parabola is very flat (somewhat similar to that for the much thicker Joukowski section of Fig. 6) and that the pressure point migration is small, wich is connected With the s-ahafen Lrewt kend of the rear end. Hari we mished to obtain a section with fixed pressure center, it would only have been necessary, after determining $c_{2}$, to draw the line BM parallel to $\sqrt{c_{1}}$, which would have hardy changed the shape of the section. We will demonstrate this by a second example.

From the practical standpoint, the considerable reduction in the thickness of the rear end of the section just considered is very noticeable. On account of the requisite thickness of the rear spar, such forms cannot ordinarily be used. We obtain another perimeter whick, to a. certain degree, realizes the opposi te extreme of retaining a nearly uniform thickness almost to the rear edge, through the following assumpiion. Still simpler than in the first example $n=2$ and our statement now reads

$$
z=\zeta+\frac{c_{1}}{\zeta} \frac{c_{2}}{\xi^{2}}
$$

Both v-values which according to (18) must be given the sum $c$, we choose as

$$
v_{1}=-\frac{c}{2} e \frac{\pi i}{3}, v_{2}=c-v_{1} .
$$

By substitution in (19) we get

$$
c_{1}=c^{2}-v_{2} v_{2}, c_{2}=-\frac{c}{2} v_{1} \quad v_{2} ; \nabla_{2}=-\frac{c^{2}}{8}(1+3 \sqrt{3 i})
$$

In a similar manner to the above, the radial line $B C=\nabla_{1} \nabla_{2}: 0$ is drawn in Fig. 13, so that $O C$ represents its length and direotion according to $-c_{1}$ : c. By bisecting the angle formed by $O C$ with the real axis, the direction of the "second axis" of the section is found, which is, now that we wish to obtain a section with a pressure center, also the direction of the "first axis." We therefore dram through $B$ the parallel to the blsector of the angle and choose on it the center $M$ of the circle $K$, with the radius $B M=a=0.5$. It is seen that $K$ incloses the endpoints $V_{1}$ and $V_{2}$ of the vectors $V_{1}, V_{2}$ and $C$. We now construct, just astin the first example, an auxiliary circle $\mathrm{K}_{1 ;}$ which contains the end points $A_{1}$ of the vectors $c_{1}: \zeta$ and derivea the end points $A^{\prime}$ of the vectors $\zeta+\frac{c_{1}}{\zeta}$ from the 12 points $A$ of the original circle $K$, with the aid of $A_{2}$. In order to obtain the ultimate section points $A^{n}$, we must still make the displacements $c_{2} \zeta^{2}$. Note here that $C B$ hasthe direction of $c_{2}$ and that from this, therefore, by the subtraction of $2 \gamma$ plus entire multiples of $60^{\circ}$, all the shifted directions are obtained. The leng $\ddagger$ : 0 of the displacements are determined either with the abacus, whereby the value $c_{2}==0.0211$ is to be used, or by calculation by means of the proportion

$$
\frac{c_{2}}{\zeta^{2}}: \frac{c_{1}}{\zeta}=\frac{c_{2}}{c_{2}}: \zeta
$$

Furthermore, it is only necessary to draw a circle with 0 for its center and the radius $\frac{c_{2}^{\prime}}{c_{i}^{\prime}}$, in order to obtain (Fig. 13) throug: HH' parallel to $O A^{\prime}$ the desired length in $A^{\prime} \mathrm{H}^{\prime}$. The pressure
center lies on the first axis at a distance of

$$
M F=c_{1}: a=\frac{c^{2}}{8} \sqrt{108}: 0.5=0.415
$$

It is seen that this section also, like the first one, has' an s-shaped curve which shows plainly on the pressure side. In contras, however, with the first exmple, the thickness of the section is much more uniform. Intermediate forms, between the two given here, can readily be obtained, by varying the value of $v$, whereby the pressure point migration can be either entirely eliminated, as in tine second, or largely, as in the first example. It is worth noting, furtrer, that both sections are almost the same with respect to the lift magnitude, since the determining ratio $4 a: b \quad(b=g r e a t e s t w i d t h)$ for the lift coefficient is, in the first case 176 : 163 and in the second case 200:183 accordingly about 1.09 in both cases. The theoretical lift coefficient is therefore $1.09 \pi$ sin $\alpha_{1}$ in which $\alpha_{1}$ is the effective angle of incidence $\alpha+\beta$. (According to Bieberoach's estimate, the coefficient is larger than $\pi \sin \alpha$, .

In the third example, a section is drawn in Fig. 14 on the same plan as the las mentioned, but with other values of $v_{2}$ and $v_{z}$, namely,

$$
v_{1}, \nabla_{2}=\frac{c}{2}(I \pm \sqrt{I+0.16 i})
$$

which might do for propeller blades. This section also has, like Fig. 13, a fixed pressure center, but shows a leaning toward the Joukowski shape. The point $V_{2}$ is still nearer 0 . In any event, Fig. in and 14 illustrate the manifold variations in shape, even
with the maintenance of the pressure-center condition $h_{0}=0$ and although we have confined ourselves to the simplest case of the general method with $n=2$. On the construction of Fig. 14 there is nothing new to say. The values of $\frac{c_{c}}{\zeta^{2}}$ were here determined: by calculation. The numerical values for the lift coefficient and the focal distance are here

$$
\begin{aligned}
c_{2} & =c(1+0.4 i), c_{2}=0.2 c^{5} 1 ; \\
\zeta A & =1.1 \pi \sin (\alpha+\beta), \quad M F=0.375
\end{aligned}
$$

If the method were carried beyond $n=2$, the attainable shapes would then become so numerous that it would be impossible to give even a glimpse of them within the narrow limits of this fundamental exposition.

Translated by N. A. C. A.


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